

Research project: SHEPHERD (UAS Standards)

Webinar: final dissemination event 23/04/24, 10:00-12:00 CET

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Welcome to this webinar!



This webinar is the final dissemination event of this research project



This project has received funding from the European Union's Horizon Europe research and innovation Programme



The EC delegated the contractual and technical management of this research action to EASA



EASA contracted Deep Blue as Consortium lead for the implementation of the research action following a public tender procedure



EASA-managed projects are addressing research needs of aviation authorities and are an important pillar of the EASA R&I portfolio



The agenda

TIME	TITLE, SPEAKER
10:00 - 10:05	Welcome to the webinar Willy Sigl, EASA
10:05 – 10:15	Research scope and objectives Natale Di Rubbo, EASA
10:15 - 11:00	Overview of the project implementation and key results Marco Ducci, Deep Blue and Alexandra Florin, Wing
11:00 - 11:10	Benefits from the project, planned follow-up actions Natale Di Rubbo, EASA
11:10 – 11:55	Questions and answers Participants, Project Team from SHEPHERD, EASA
11:55 – 12:00	Concluding remarks Willy Sigl, EASA

Note: this webinar will be recorded and made available at the EASA website after the event.



Question and Answers

- → For sending questions and comments, please use the slido app, which is also accessible through WebEx:
 - www.slido.com
 - event code: 1560086







Research Scope and Objectives

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Standards for drone operations

How to comply with the EU drone regulation?













Overview of the project implementation and key results

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SHEPHERD Consortium

- Project Coordinator
- Technical Lead
- Partners
- Subcontractors





SHEPHERD Stakeholders Consultation Group (SCG)

\rightarrow Role

- → Provide review, recommendations and feedback on project activities and findings;
- → Coordinate with standards development efforts currently on-going.

\rightarrow **Composition** – 10 experts from:

- \rightarrow <u>SDOs</u>: EUROCAE, ASTM, ASD-STAN
- → <u>Associations</u>: GUTMA, JEDA
- → <u>Authorities/Organisations</u>: CASA, EUROCONTROL



SHEPHERD Objectives

- → Define a methodology for the technical assessment of UASrelated standards
- → Identify which standards (or elements thereof) are considered technically adequate to demonstrate compliance with the requirements within the scope of the SHEPHERD project
- → Identify which standards (or elements thereof) are <u>NOT</u> considered technically adequate to demonstrate compliance with the requirements within the scope of SHEPHERD project.



SHEPHERD Structure

WP 4 – Communication, dissemination, knowledge-sharing, stakeholder management



WP2 – Identification of satisfactory industry standards

WP3 – Justification for not acceptable industry standards

WP0 – Project management



SHEPHERD Timeline





Where to find SHEPHERD deliverables

\rightarrow EASA's dedicated webpage

https://www.easa.europa.eu/en/rese arch-projects/shepherd-uas-standards





SHEPHERD Methodology: 4 steps

- → Step #1 Identify the standards in scope and the requirements against which the standards need to be assessed;
- → Step #2 Categorise the requirements against which the standards in scope need to be assessed;
- → Step #3 Assess with a 4-eye independent principle each proposed standard; and
- → **Step #4** Summarise the assessment.



Step #1 – Definition of SHEPHERD scope

→ Requirements (+550 in total)

- → <u>SORA</u>: Operational Safety Objectives (OSOs); Ground risk mitigations; Tactical mitigations performance requirements (TMPRs); and Containment requirements;
- → EASA Special Condition (SC-) Light UAS for Medium (M) and High (H) risk: SAIL III to VI;
- → <u>U-Space services</u> regulatory requirements and associated AMC & GM.

→ Standards (47 in total)

- \rightarrow All standards identified by 'AW-Drones' project as adequate.
- → Any additional standards part of EUROCAE WG-105, ASTM F38 and RTCA SC-228/SC-147 roadmaps and deemed to be published in the timeline of the SHEPHERD's project.



Step#2 – Categorisation of requirements

The requirements identified in **Step#1** are categorised as either:

- → A.1: Technical objective-based: performance-oriented requirements targeting a specific technical consideration or design but leaving flexibility on the implementation up to the UAS manufacturer.
- → A.2: Operational / organisational objective-based: performance-oriented requirements leaving flexibility on the implementation at operational (e.g., with respect to training / manual and procedures) or organisational levels up to the UAS operator.
- → B: Technology-dependent: performance requirements whose implementation strongly dependent on the technology chosen by the applicant.



Step#3 – Assessment

Criteria	Standard assessment result						
C1 – Level of confidence that the standard	L – Level of confidence that the standard meets the requirement						
C1.1 – Completeness / coverage	 C: Completely addresses the requirement assessed Pa: partially addresses the requirement assessed N/A: does not address the requirement assessed 						
C1.2 – Correctness	 A: Applicable as it is I: Intent of the requirement can be applied, provided some specific (slight) adaptations are made T: to be Tailored 						
C1.3 – Proportionality	 Pr: the standard is proportionate to the associated SAIL(s) No Pr: the standard is not proportionate to the associated SAIL(s). 						
C1.4 – Interoperability	 Interoperable Not interoperable N/A 						
C2 – Easiness of implementation of the sta	ndard						
C2.1 – Proven implementability / maturity	 Proven to be implementable / mature Not proven to be implementable / mature 						
C2.2 – Implementation agnostic	 Implementation agnostic Not implementation agnostic 						



Step #4 – Summary of the assessment

View per standard	 One excel file per standard with detailed technical assessment One word file per standard summarising the assessment
View per requirement	 One excel file listing, for all requirements, the recommended standards or section / subsection / paragraph(s) thereof



SHEPHERD Results: view per standard (1)



Standard	Retained requirement	All levels?
	Light-UAS.23xxx	SAIL III to VI
Std XXXX-20YY	Light-UAS.24yy	SAIL III to VI
	Light-UAS.25zz	SAIL III to VI
	OSO#XX	Integrity; SAIL I to VI
	OSO#YY, Criterion #z	SAIL II to VI

Sections of the standard	Detailed assessment scope	Rationale for exclusion from detailed assessment			
Introduction	N/A	General introduction to the standard			
1. Scope	N/A	High-level			
2. References	N/A	Reference to other relevant standards			
3. Terms and definitions	N/A	List of terms and abbreviations used throughout the standard			
4. Procedures for the safe []		Light-UAS.23xx: design and construction not addressed			
4.1 Normal procedures	OSO#YY, Criterion #z (Integrity)	<u>Light-UAS.24yyy</u> : lift / thrust / power system installation not addressed <u>Light-UAS.25zz</u> : systems and equipment not addressed			
4.2 Off-nominal procedures		<u>OSO#XX (Integrity)</u> : remote crew training / qualification not addressed <u>OSO#YY, Criterion #z (Assurance)</u> : audits or verifications not addressed			
5. Design and configuration of […] functionality	Light-UAS.23xx	<u>Light-UAS.24yyy</u> : lift / thrust / power system installation not addressed <u>Light-UAS.25zz</u> : systems and equipment not addressed <u>OSO#XX (Integrity)</u> : remote crew training / qualification not addressed <u>OSO#YY, Criterion #z</u> : procedures for the safe [] not addressed			

SHEPHERD Results: view per standard (2)



Standard					Requirement		Level of	Associated Requireme	Requirement	Deta	ent								
	Section	oso	Integrity / Assurance	Criterion	robustness	SAIL	type	Criteria	Technical assessment result	Comments / rationale	Overall rating								
								C1.1 completeness / coverage	Partially addresses the requirement										
								C1.2 correctness	Intent ca										
								C1.3 proportionality	Proportionate to the SAIL										
Std XXXX-20YY	. 4.1				Low (L)	1&11	&	&	&	1&11	&	&	1&11	1&11	&	&	C1.4 interoperability	N/A	
							A.2. operational / organisational	C2.1 implementability / maturity	Proven to be implementable		_								
		#YY	Integrity	ity #z				C2.2 implementation agnostic	Yes										
						∕ledium (M) III & IV		C1.1											
								C1.2			Deserves de d								
					Modium (M)			C1.3											
								C1.4			Recommended								
								C2.1											
								C2.2											
					High (H)	IV & V					Not recommended								



SHEPHERD Results: view per standard (3)

Standard XXXX-20YY							
Requirement	SAIL Integrity / Assurance	Recommended section(s)	Additional relevant information				
OSO#XX	Integrity SAIL III	Sections a.bc & d.e	Partial coverage				
OSO#XX	Integrity SAIL IV						
OSO#XX	Integrity SAIL V & VI	Sections f.gh, i.jk & I.m	Full coverage				
Light-UAS.23xxx	SAIL III & IV						
Light-UAS.23xxx	SAIL V & VI						

Standard XXXX-20YY							
Section	Title / Subject	Requirement	Required tailoring / complementing				
Section n.op	Power system performance	Light-UAS.24yy Medium (M) & High (H)	No criteria, limitations, or instructions are provided				
Section q.r	Equipment installation	Light-UAS.25zz Medium (M)	The proposed criteria is deemed excessive for Medium (M)				

ZEASA D2.1-D3.1 Identification of satisfactory industry standards and justification for not acceptable industry standards An Agency of the European U

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Reference	Retained requirement	All levels?	Reference	Retained requirement	All levels?
	SC Light-UAS.2100	SAIL III to VI		SC Light-UAS.2400	SAIL III to VI
	SC Light-UAS.2102	SAIL III to VI		SC Light-UAS.2405	SAIL III to VI
	SC Light-UAS 2105	SAIL III to VI		SC Light-UAS.2410	SAIL III to VI
	SC Light LIAS 2135	SAIL III to VI		SC Light-UAS.2415	SAIL III to VI
				SC Light-UAS.2430	SAIL III to VI
	SC Light-UAS.2160	SAIL III to VI		SC Light-UAS.2510	SAIL III to VI
	SC Light-UAS.2235	SAIL III to VI	ASTM F3298 - 19 Standard Specification for	SC Light-UAS.2529	SAIL III to VI
	SC Light-UAS.2240	SAIL III to VI	Design, Construction, and	SC Light-UAS.2530	SAIL III to VI
ASTM 52209 - 10	SC Light-UAS.2250	SAIL III to VI	Verification of Lightweight	SC Light-UAS.2605	SAIL III to VI
Standard Specification for	SC Light-UAS.2260	SAIL III to VI (UAS)		SC Light-UAS.2610	SAIL III to VI
Design, Construction, and	SC Light-UAS,2300	SAIL III to VI		SC Light-UAS.2615	SAIL III to VI
Verification of Lightweight	CC Light LIAC 2205			OSO#02 (Integrity)	SAIL III to VI
Unmanned Aircraft Systems	SC Light-DAS.2305	SAIL III IO VI		OSO#05	SAIL III to VI
(BAS)	SC Light-UAS.2325	SAIL III to VI		OSO#06	SAIL II to VI
	SC Light-UAS.2340	SAIL III to VI		OSO#13	SAIL I to VI
	SC Light-UAS.2370	SAIL III to VI		OSO#20	SAIL II to VI
	SC Light-UAS.2375(a)	SAIL III to VI			
	SC Light-UAS.2375(b)	SAIL III to VI			
	SC Light-UAS.2380(a)	SAIL III to VI		V	
	SC Light-UAS.2380(b)	SAIL III to VI			
	SC Light-UAS.2380(c)	SAIL III to VI			





Sections of the standard	Detailed assessment scope	Rationale for excluding some requirements from the assessment of specific paragraphs
1. Scope	N/A	This section provides the scope of the document. Units of measure are expressed in imperial units and this may be an issue for implementation in the EU.
2. Referenced Documents	N/A	Title only
2.1 ASTM Standards	N/A	Only reference documents listed
2.2 ANSI Standard	N/A	Only reference documents listed
2.3 FAA Standard	N/A	Only reference documents listed
2.4 Federal Standard	N/A	Only reference documents listed
2.5 Joint Authorities for Rulemaking of Unmanned Systems	N/A	Only reference documents listed
2.6 Unmanned Systems Canada Best Practices	N/A	Only reference documents listed
3. Terminology	N/A	Title only
3.1 Definitions of Terms Specific to This Standard	N/A	Definition of terms specific to the standard
3.2 Abbreviations	N/A	List of abbreviations
4. Significance and Use / Applicability	N/A	Intent of the Specification / Standard
5. Flight	N/A	Title only

		Generic proof of compliance requirements.
		FLIGHT
		Light-UAS.2100: No details are provided as regards the method or conditions to determine
		the limits of the centre of gravity.
		Light-UAS.2105(b)(d)(e): Data to be included in the flight manual is only addressed at high
		level (2105(b)), there is no explicit reference or evidence that the procedures used for
		determining performance are executable consistently in atmospheric conditions expected to
		be encountered in operation and by a remote crew of average skill (2105(d)), Losses due to
		atmospheric conditions etc. are not considered in the definition of the performance.
		Light-UAS.2135: Controllability, manoeuvrability, and stability are not addressed.
		Light-UAS.2160: Vibration and buffeting are not addressed.
		STRUCTURES
		Light-UAS.2235: Structural strength and deformation are not addressed.
		Light-UAS.2240: Structural durability is not addressed.
		Light-UAS.2250: Design and construction principles are not addressed.
		Light-UAS.2260: Materials and manufacturing processes are not addressed.
		DESIGN AND CONSTRUCTION
		Light-UAS.2300: UA flight control systems are not addressed.
		Light-UAS.2305: Landing gear systems are not addressed.
		Light-UAS.2325: Fire protection is not addressed.
		Light-UAS.2340: Design and construction information is not addressed.
		Light-UAS.2370: Transportation, assembly, reconfiguration, and storage are not addressed.
		Light-UAS.2375(a): Payload installation / accommodation is not addressed.
		Light-UAS.2375(b): Payload operational limitations, procedures, and instructions are not
5.1 Droof of	Light LLAS 2102	addressed.
Compliance	Light UAS 2105(a)8(a)	Light-UAS.2380(a): Ancillary equipment performance and design are not addressed.
compliance	Light-OA3.2 105(a)a(c)	Light-UAS.2380(b): Ancillary equipment instructions, information, and limitations as regards
		the interface with the UA are not addressed.
		Light-UAS.2380(c): UA design to operate safely using the ancillary equipment is not
		addressed.
		LIFT/THRUST/POWER SYSTEM INSTALLATION
		Light-UAS.2400: Lift/Thrust/Power systems installation is not addressed.
		Light-UAS.2405: Lift/Thrust/Power system integrity is not addressed.
		Light-UAS.2410: Lift/Thrust/Power endurance and durability are not addressed.
		Light-UAS.2415: Lift/Thrust/Power calibration, ratings, and operational limitations are not
		addressed.
		Light-UAS.2430: Energy storage and distribution systems are not addressed.
		SYSTEMS AND EQUIPMENT
		Light-UAS.2529: UAS navigation function is not addressed.
		Light-UAS.2530: UA external lights are not addressed.
		REMOTE CREW INTERFACE AND OTHER INFORMATION
		Light-UAS.2605: Command unit installation and operation information are not addressed.
		Light-UAS.2610: Instrument markings, control markings, and placards are not addressed.
		Light-UAS.2015: Flight, havigation, and thrust/lift/power system instruments are not
		addressed.
		UPERATIONAL SAFETY OBJECTIVES (OSOS)
		0.000#02. UAS manulacturing is not addressed.
		0.000000 CAS design considering system safety and reliability is not addressed.
		USU#00. US IIIIK periormance is not addressed.
		OSO#15. External services supporting UAS operations are not addressed.
1	1	1050#20. Human machine Interface (HNII) evaluation is not addressed.





			Level of	SAIL	Requirement type (A.1 or A.2)	Detailed assessment			D. C.	Additional
standard	Section	Light-UAS	robustness			Criteria	Technical assessment	Comments / rationale	Rating	information
		5.1 Light-UAS. 2105(a)&(c)		n (M) III & IV	A.2. Operational / organisational	C1.1 - Coverage	Pa: partially addresses the requirement	This section defines the speed range that shall be considered when testing the configuration and determining the performance limits. This does not fully define the flight envelope limitation defined by the requirement.	Recommended	Recommended in combination with sections 6.1 - 6.5 or 6.6.2 - 66.6 depending on UAS configuration
	5.1					C1.2 - Correctness	I: Intent of the requirement can be applied w/ specific (minor) adaptations	It needs to be amended to mention the need to demonstrate the performance in still air and standard atmospheric conditions at sea level		
			Medium (M) nt-UAS. 5(a)&(c)			C1.3 - Proportionality	Pr: proportionate to the targeted SAIL(s)	The speed limits are expressed as a function of the demonstrated flight diving speed so the guidance is assessed as proportionate.		
ASTM						C1.4 - Interoperability	N/A			
F3298-19						C2.1 - Proven implementability	Proven to be implementable	Considered as well-known and used standards and procedures.		
							C2.2 - Implementation agnostic	Yes	It is not restricting concerning the way the limits are implemented	
						C1.1	Same as Medium (M)		Recommended in
						C1.2	Same as Medium (M)		
						C1.3	Same as Medium (M)		combination with
			High (H)	V & VI		C1.4	N/A		Recommended	or 6.6.2 - 6.6.6
						C2.1	Not proven to be implementable / mature	There is no evidence that this standard has been used for UAS operated at SAIL V and VI		depending on UAS configuration
						C2.2	Same as Medium (M)		



3. Recommended sections

This section provides the list of recommended sections, subsections, paragraphs, or combination thereof of ASTM F3298-19 that may be used as a basis for a MoC for the requirements or a part thereof as identified in the detailed technical assessment, which can be accessed <u>here</u>.

ASTM F3298-19						
Requirement	Related SAIL Integrity / assurance	Recommended section(s), subsection(s), paragraph(s), or combination thereof	Additional relevant information			
Light-UAS.2105(a)&(c)	SAIL III to VI	5.1 5.4 5.6.1 6.1 - 6.5 6.6.2 - 6.6.6	The recommended sections only partially address Light-UAS.2105(a)&(c) since they only cover specific configurations and not all elements to demonstrate safe operations within the operational envelope. The following minor adaptations are proposed: - the standard should also include the rate of descent, max hover altitude, approach speed, and external load performances as required by Light-UAS.2105; - Sections 5.1, 6.1 - 6.5, and 6.6.2 - 6.6.6 should indicate the need to demonstrate the performance in still air and standard atmospheric conditions at sea level; - Sections 6.2 - 6.6.6 should include guidance on how to define performance in hovering flight conditions.			



4. Non-recommended sections

This section provides the list of elements of ASTM F3298-19 that need to be tailored and/or complemented before being proposed as a MoC for the requirements or a part thereof as identified in the detailed technical assessment, which can be accessed here.

ASTM F3298-19							
Section, subsection, or paragraph to be tailored / complemented	Title / subject	Requirement and SAIL	Required tailoring / complementing				
5.1	Proof of compliance	Light-UAS.2102 SAIL III – Medium (M)	Although Section 5.1 provides additional details on the definition of the upper flight speeds that create the upper boundary of the flight envelope, the requirements proposed are excessive for UAS intended to be used in a SAIL III operation.				





SORA reference SC Light-UAS Subpart U-space service	Integrity / Assurance Light-UAS.2xxx U-space article	Level of robustness Associated AMC & GM	Criterion #	Recommended section(s)	Coverage	Overall coverage
	late avity	Low (L)	N/A	Std XXXX-20YY: Sections a.aa & b.b Partial		F
OSO#XX	integrity			Std ZZZ-20yy: Section c	Partial	Fuii
	A		Criterion #1			
	Assurance	LOW (L)	Criterion #2			
	Integrity	Medium (M)	N/A			
	A		Criterion #1			
	Assurance	wealum (w)	Criterion #2			
	Integrity	High (H)	N/A			
	A		Criterion #1			
	Assurance	High (H)	Criterion #2			



SHEPHERD Results: view per requirement - example

	SORA reference	Integrity / Assurance	Level of							
	SC-Light-UAS Subpart	Light-UAS .2xxx	robustness	robustness Associated	tness Criterion #	Requirement full reference	Recommended section(s)	Coverage	Overall coverage	Comments
	U-Space service	U-Space article	AMC & GM							
	SORA OSOs									
						ED-279 Partial		The overall coverage is considered 'Full', as ED-28 provides a generic methodology to perform safety		
	OSO#05	Integrity	Low (L)	N/A	OSO#05 Integrity	ED-280	Full	Full	assessment suitable for OSO#5 Low Integrity. For their part, ED-279 and the recommended sections of F3298-19 complement ED-280. Specifically, ED-279 provides a methodology to perform an FHA as a starting point, whilst F3298- indicates a way to handle specific failure cases.	
					Low (L)	F3298-19: Sections 5.6.2, 7.9.2.7, 7.10.1.2(5), 16.3 & A2.3	Partial			
Γ					0004051	<u>ED-279</u>	Partial		Recommended sections of F3309 complete ED-2	
	OSO#05	Assurance Low (L)	N/A	Assurance	ED-280	Partial	Full	by providing criteria for design and installation appraisal. ED-279 provides a methodology to perform an FHA as a starting point.		
				LOW (L)	F3309/F3309-21: Sections 4.4.1, 4.4.2 & 4.6	Partial				



Summary and Conclusions





for 10% no standard is required



Benefits from the project, planned follow-up activities

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Assessment of standards in support of SORA

- → Develop NPA related to RMT 0730 (regular update of AMC and GM to Regulations (EU) 2019/945 and 2019/947)
 - \rightarrow Planned for Q1 2025
- → Publish Decision amending AMC and GM to Regulation (EU) 2019/947
 - \rightarrow Planned for Q1 2026





Assessment of standards in support of SC light UAS

→ Develop Means of Compliance (MOC)

- → MoC Light-UAS.2405 Lift/Thrust/Power system integrity published in Q3/23
- → MoC Light-UAS.2410 Lift/Thrust/Power system endurance and durability published in Q3/23
- → MoC 2575(a) command, control and communication contingency, publication planned for Q2/24





Assessment of standards in support of U-space

- → Develop NPA related to RMT 0748 (regular update of AMC and GM to Regulations (EU) 2021/664, 665, 666)
 - \rightarrow Phased and iterative approach
 - \rightarrow Publication planned <u>in</u> 2025
- → Publish Decision amending AMC and GM to Regulation (EU) 2021/664
 - \rightarrow To follow

Certain AMC/GM are planned to be updated which may induce local need of reassessment





ÉUSCG (کچک) EUSCG

Update of Rollout development plan

For those standards not recommended, propose to standardisation bodies to amend standards to become fully compliant

Coordinate the activities of standardisation bodies to fill the gaps





Questions and answers

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Question and Answers

- → For sending questions and input, please use the slido app, which is also accessible through WebEx:
 - www.slido.com
 - event code: 1560086







Concluding Remarks

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A+8= 2+1+2 31×4





Thank you for joining this webinar!

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