



TYPE CERTIFICATE DATA SHEET

No. EASA.IM.R.507

for

R66

Type Certificate Holder

Robinson Helicopter Company

2901 Airport Drive

Torrance, CA 90505

U.S.A.

For Model: R66



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SECTION 1: R66I. General

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|--|--|
| 1. Type/ Model/ Variant | |
| 1.1 Type | R66 |
| 1.2 Model | R66 |
| 1.3 Variant | n/a |
| 2. Airworthiness Category | Small Rotorcraft, Category B |
| 3. Manufacturer | Robinson Helicopter Company
2901 Airport Drive
Torrance, California 90505, USA |
| 4. Type Certification Application Date | to FAA: 6 September 2006
to EASA: 19 May 2010 |
| 5. State of Design Authority | FAA |
| 6. Type Certificate Date by NAA | by FAA: 25 October 2010 |
| 7. Type Certificate n° | by FAA: R00015LA-R |
| 8. Type Certificate Data Sheet n° | by FAA: R00015LA-R |
| 9. EASA Type Certification Date | 30 April 2014 |

II. Certification Basis

- | | |
|---|--|
| 1. Reference Date for determining the applicable requirements | 8 May 2009 |
| 2. Airworthiness Requirements | CS 27, Amdt. 2, dated 17 November 2008 |
| 3. Special Conditions | none |
| 4. Exemptions | none |
| 5. Deviations | none |
| 6. Equivalent Safety Findings | CS 27.695(a)(1) Power boost and power-operated control system (CRI D-01) |
| 7. Requirements elected to comply | none |
| 8. Environmental Protection Requirements | See EASA Type Certificate Data Sheet for Noise TCDSN EASA.IM.R.507 |
| 9. Operational Suitability Data (OSD) | see SECTION 2 below |

III. Technical Characteristics and Operational Limitations

- | | |
|---------------------------|--|
| 1. Type Design Definition | R66 Master Drawing List (MDL) 0066 |
| 2. Description | Main rotor: 2-blade, free to teeter and cone, rigid in-plane
Tail rotor: 2-blade, free to teeter, rigid in-plane
Fuselage: Riveted aluminium sheet and welded steel tube for primary structure, fiberglass & thermoplastic |



for secondary structure. Seats integral to cabin structure.

Landing gear: Aluminium skids
Powerplant: Single turbine
Avionics: Analogue or EFIS

3. Equipment

Basic equipment must be installed and operational prior to registration of the helicopter.
Optional equipment per RHC drawing F025.

4. Dimensions

4.1 Fuselage	Length:	11.66 m
	Width hull:	1.47 m
	Height:	3.48 m
4.2 Main Rotor	Diameter:	10.06 m
4.3 Tail Rotor	Diameter:	1.52 m

5. Engine

5.1 Model	Rolls-Royce 1 x 250-C300/A1
5.2 Type Certificate	FAA TC/TCDS n°: E4CE EASA TC/TCDS n°: EASA.IM.E.052

5.3 Limitations

5.3.1 Installed Engine Limitations and Torque Limits

	TQ limits [% (hp)]	Gas generator N ₁ [rpm (%)]	PWR turbine N ₂ [rpm (%)]	Temperature MGT [°C]
TOP (5 min)	100 (270)	53 519 (105)	6 076 (101)	782
MCP	83 (224)	53 519 (105)	6 076 (101)	706
Max. starting	---	---	---	927*
<u>Note:</u> *10 second limit above 782°C				

5.3.2 Transmission Torque Limits

	Max. TQ [Nm]	PWR turbine N ₂ [%]
TOP (5 min)	320	101*
MCP	266	101
<u>Note:</u> *100% = 6 016 rpm		

6. Fluids (Fuel/ Oil/ Additives)

6.1 Fuel Jet A or Jet A-1 conforming to ASTM D 1655,
Jet B conforming to ASTM D 6615,
JP-4 or JP-5 conforming to MIL-DTL-5624,
JP-8 conforming to MIL-DTL-83133

6.2 Oil Engine: AS 5780 HPC
MRGB/TRGB: Robinson P/N A257-22



6.3 Additives

Anti-icing additive conforming to MIL-DTL-85470 must be added to Jet A, Jet A1, or Jet B when ambient temperature is below 4°C. Check with fuel supplier to determine if supply includes additive. If not, add per manufacturer's instructions.

7. Fluid capacities

7.1 Fuel

Fuel tank capacity: 282 litres (74.6 US gal)
Usable fuel: 279 litres (73.6 US gal)

7.2 Oil

Engine: 5.7 litres (1.5 US gal)
MRGB: 1.9 litres (2 qt)
TRGB: 0.10 litres (0.11 qt)
Hydraulic reservoir: 0.62 litres (0.65 qt)

7.3 Coolant System Capacity

n/a

8. Air Speed Limitations

Take-off Gross Weight	PWR on V_{NE} [KIAS]	PWR off V_{NE} [KIAS]
Less than 998 kg	140	100
998 to 1 225 kg, or Airborne Observation Helicopter version (any gross weight), or Pop-out Floats version (floats stowed, any gross weight)	130	100

Notes:

- MSL V_{NE} values shown above.
- For reduction of V_{NE} with altitude and temperature, see R66 Pilot's Operating Handbook and EASA- approved RFM (RTR 661).
- Airspeed limit is 65 KIAS for power settings above 83% torque.
- Airspeed limit is 100 KIAS for any combination of doors off.

9. Rotor Speed Limitations

Condition	Minimum		Maximum	
	[rpm*]	[%]	[rpm*]	[%]
Power on	404	99	412	101
Power off	359	88	432	106

Note: *Main Rotor

10. Maximum Operating Altitude and Temperature

10.1 Altitude

14 000 ft (4 270 m) DA

10.2 Temperature

From -40°C to ISA+35°C, limited to +50°C



11. Operating Limitations

VFR day and night
Non-icing conditions

12. Maximum Mass

1 225 kg

13. Centre of Gravity Range

Gross weight [kg]	Longitudinal C.G.	
	FWD limit [mm]	AFT limit [mm]
635	2 311	2 604
1 043	---	2 604
1 134	2 311	---
1 225	2 337	2 489
Longitudinal C.G. [mm]	Lateral C.G.	
	Left limit [mm]	Right limit [mm]
2 311	-89	+89
2 540	-89	+89
2 604	-38	+38

14. Datum

Longitudinal:
the datum plane (STA 0) is located at 2 540 mm
(100 in) forward of main rotor centreline.

Lateral:
fuselage median plane

15. Levelling Means

Refer to R66 Maintenance Manual, and,
Instructions for Continued Airworthiness (RTR 660),
Chapter 8

16. Minimum Flight Crew

1 pilot (right seat)

17. Maximum Passenger Seating Capacity

4

18. Passenger Emergency Exit

4, two on each side of the passenger cabin
(intended for normal use)

19. Maximum Baggage/ Cargo Loads

Maximum mass: 136 kg (300 lb)
Maximum loading: 244 kg/m² (50 lb/ft²)
Underseat baggage compartments:
Maximum mass: 23 kg (50 lb)
For any seat location, the maximum combined
weight of the load on the seat (e.g. occupant) plus
the weight of stowed items and any installed
equipment in the underseat baggage compartment
is 136 kg (300 lb).

20. Rotor Blade Control Movement

Main Rotor:

Collective pitch	13.0° ±0.5° total travel	
Cyclic pitch	forward	13.50° to 14.25°
	aft	13.50° to 14.25°
	left	7.5° to 8.5°
	right	6.0° to 7.0°



Tail Rotor:

Collective pitch	left pedal	15.5° to 16.5°
	right Pedal	18.5° to 19.0°

21. Auxiliary Power Unit (APU)

none

22. Life-limited Parts

See Robinson Maintenance Manual and Instructions for Continued Airworthiness (RTR 660).

Retirement times are listed in the EASA-approved "Airworthiness Limitations" section of Chapter 4, dated 25 October 2010, or later revisions.

IV. Operating and Service Instructions

1. Flight Manual

R66 Pilot's Operating Handbook and EASA-approved Rotorcraft Flight Manual, RTR 661, dated 25 October 2010, with revisions through 26 November 2013, or later.

2. Maintenance Manual

R66 Maintenance Manual and Instructions for Continued Airworthiness (RTR 660 Volume I).

3. Structural Repair Manual

none

4. Weight and Balance Manual

none

5. Illustrated Parts Catalogue

R66 Illustrated Parts Catalog (RTR 660 Volume II)

6. Service Letters and Service Bulletins

R66 Service Letters and Service Bulletins as published by Robinson Helicopter Company.

7. Required Equipment

The basic required equipment as prescribed in the applicable airworthiness regulations (see Certification Basis) must be installed in the aircraft for certification, or as required by the Master Minimum Equipment List. In addition, the EASA-approved Rotorcraft Flight Manual is required (see Flight Manual)



V. Notes

1. Manufacturer's eligible serial numbers:
s/n 0560 and subsequent, or s/n 0004 thru 0559 with R66 Service Letter SL-08 completed.
2. Instrument markings:
Any cockpit instruments installed by a third party must be marked with limit markings and range markings in accordance with Robinson's marking scheme.
3. *deleted*
4. Noise configuration:
The "Clean" and "Dirty" configurations for noise characteristics are defined in the EASA-approved Rotorcraft Flight Manual, Section 5.
5. Designation:
R66 Turbine is used as marketing designation for the basic R66 helicopter. R66 Turbine Marine is used as a marketing designation for the R66 with optional pop-out floats.



SECTION 2: OPERATIONAL SUITABILITY DATA (OSD)

The OSD elements listed below are approved by the European Aviation Safety Agency as per Commission Regulation (EU) 748/2012, as amended by Commission Regulation (EU) No 69/2014.

I. OSD Certification Basis

- I.1 Reference Date for determining the applicable OSD requirements
12 August 2014
- I.2 MMEL - Certification Basis
Special Condition SC-CS-GEN-MMEL-H, Initial Issue
- I.3 Flight Crew Data - Certification Basis
CS-FCD, Initial Issue

II. OSD Elements

- II.1 MMEL
EASA MMEL for R22, R44, and R66, Appendix 1 to RTR 666, dated 17 November 2015,
or subsequent approved revisions
- II.2 Flight Crew Data
RTR 665, EASA Operation Suitability Data, Flight Crew Data, Initial OSD Issue,
or subsequent approved revisions



SECTION: ADMINISTRATIVEI. Acronyms and Abbreviations

AFT	Aft	n/a	Not applicable
C.G.	Centre of Gravity	OSD	Operational Suitability Data
CRI	Certification Review Item	PA	Pressure Altitude
CS	Certification Specification	PWR	Power
DA	Density Altitude	RHC	Robinson Helicopter Company
DP	Datum Point	RFM	Rotorcraft Flight Manual
EFIS	Electronic Flight Information System	RTR	Robinson Technical Report
FAA	Federal Aviation Administration	s/n	Serial Number
FCD	Flight Crew Data	SC	Special Condition
FWD	Forward	STA	Station
ISA	International Standard Atmosphere	TOP	Take-Off Power
KIAS	Knots Indicated Air Speed	TRGB	Tail Rotor Gearbox
max	Maximum	TQ	Torque
MC	Maximum Continuous	VFR	Visual Flight Rules
MCP	Maximum Continuous Power	VFR	Visual Flight Rules
MGT	Measured Gas Temperature	V _{NE}	Never Exceed Speed
MMEL	Master Minimum Equipment List		
MRGB	Main Rotor Gearbox		
MSL	Mean Sea Level		

II. Type Certificate Holder Record

Type Certificate Holder	Period
Robinson Helicopter Company 2901 Airport Drive Torrance, California 90505, USA	Since 25 October 2010

III. Change Record

Issue	Date	Changes	TC issue
Issue 01	30 Apr 2014	Initial issue of EASA TCDS	Initial Issue, 30 April 2014
Issue 02	11 Dec 2015	OSD section added	---

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