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# TYPE-CERTIFICATE DATA SHEET

NO. EASA.A.599

for

**ASG 32**

Type Certificate Holder

**Alexander Schleicher GmbH & Co. Segelflugzeugbau**

Alexander-Schleicher-Str. 1  
36163 Poppenhausen  
Germany

For model: ASG 32  
ASG 32 EI  
ASG 32 Mi



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**Section A: ASG 32**

**A.I General**

- |   |  |
|---|--|
| 1. Type/ Model/ Variant                     |  |
| 1.1 Type:                                   | ASG 32   |
| 1.2 Model:                                  | ASG 32   |
| 2. Airworthiness Category                   | Sailplane, CS-22 - Utility                       |
| 3. Manufacturer                             | Alexander Schleicher GmbH & Co. Segelflugzeugbau |
| 4. EASA Type Certification Application Date | 01 November 2012                                 |
| 5. EASA Type Certification Date             | 11 Februray 2016                                 |

**A.II EASA Certification Basis**

- |   |   |
|---|---|
| 1. Reference Date for determining the applicable requirements |   |
| 2. Airworthiness Requirements                                 | Certification Specification for Sailplanes and Powered Sailplanes, issued 24. September 2008 (CS-22, Amdt. 2) |
| 3. Special Conditions   | EASA SC A.22.1-01 - 850 kg MTOM for variant ASG 32  |
| 4. Exemptions   | None  |
| 5. (Reserved) Deviations                                      | None  |
| 6. Equivalent Safety Findings                                 | CS 22.335 (f) calculation of $V_D$ according to OSTIV<br>CS 22.585(a) reduced by factor 1.2                   |
| 7. Environmental Protection                                   | None  |



### **A.III Technical Characteristics and Operational Limitations**

1. Type Design Definition List of the drawing files ASG 32, issue 31.01.2016
2. Description Double-seat, mid-wing CRP/GRP/ARP-composite construction for FAI 20m class with flaps, double-panel Schempp-Hirth airbrakes on upper wing surface, winglets, water ballast tanks in the wing and optional in the fuselage, retractable landing gear with hydraulic disc brake, T-shaped horizontal tail (fixed horizontal stabilizer with elevator) fin and rudder.
3. Equipment  
Min. required Equipment:  
1 Air speed indicator (up to 300 km/h) (front seat)  
1 Altimeter (front seat)  
1 Outside air temperature gauge (front seat)  
2 4-Point safety harness (symmetrical)  
2 Parachute or Cushion for back rest (~ 8cm thickness)  
Additional Equipment refer to Flight and Maintenance Manual
4. Dimensions  
Span: 20,00 m  
Wing area: 15,70 m<sup>2</sup>  
Length: 9,07 m  
Height: 1,84 m
5. Launching Hooks  
Nose tow hook "E 22", LBA Datasheet No. 11.402/9 NTS  
Safety hook „Europa G 88“, LBA Datasheet No. 60.230/2
6. Weak Links  
Ultimate Strength:  
- for winch- and auto-tow launching max. 1100 daN  
- for aero-tow max. 1100 daN
7. Load Factors  
+5,3 / -2,65 (up to V<sub>A</sub>)  
+4,0 / -1,5 (up to V<sub>NE</sub>)
8. Air Speeds  
Manoeuvring Speed V<sub>A</sub> 180 km/h  
Never Exceed Speed V<sub>NE</sub> 270 km/h  
Maximum permitted Speeds  
- with flaps at 1, 2, 3, 4 V<sub>FE</sub> 270 km/h  
- with flaps at 5,6 V<sub>FE</sub> 180 km/h  
- with flaps at L V<sub>FE</sub> 150 km/h  
- in rough air V<sub>RA</sub> 180 km/h  
- for winch launching V<sub>W</sub> 140 km/h  
- for aerotowing V<sub>T</sub> 180 km/h  
- for gear operation V<sub>LO</sub> 180 km/h



9. Approved Operations Capability	VFR-Day Cloud flying not permitted Aerobatic manoeuvres not permitted, except Spinning
10. Launch methods	Aerotow Winch and Auto-Tow
11. Maximum Masses	Max. Mass: 850 kg Max. Mass of Non-Lifting Parts: 550 kg
12. Centre of Gravity Range	156 mm – 385 mm aft of datum
13. Datum	Wing leading edge at root rib
14. Levelling Means	Slope 1000 : 27 placed on upper side of fuselage boom horizontal
15. Control Surface Deflections	Refer to Maintenance Manual
16. Minimum Flight Crew	1
17. Maximum Passenger Seating Capacity	1
18. Baggage/ Cargo Compartments	9 kg
19. Lifetime limitations	Refer to Maintenance Manual



#### **A.IV Operating and Service Instructions**

- |   |   |
|---|---|
| 1. Flight Manual  | Flight Manual ASG 32, issue 01.12.2015, or later EASA approved revisions                          |
| 2. Maintenance Manual                                   | Maintenance Manual ASG 32, issue 15.01.2016, or later revisions                                   |
| 3. Structural Repair Manual                             | General Repair Manual for Alexander Schleicher Sailplanes and Powered Sailplanes, latest revision |
| 4. Manual for the TOST Release, latest approved version |   |



## **A.V** Notes

1. Manufacturing is confined to industrial production.
2. The surface colour of all fibre reinforced parts, which are exposed to sun radiation, must be painted either in
  - White
  - RAL 2004 (Reinorange)
  - RAL 2009 (Verkehrsorange)
  - RAL 3020 (Verkehrsrrot)
  - or other colours listed in the maintenance manual section 13.4, maintenance instruction “coloured surfaces”Exceptions are the areas for markings and registration, engine bay and cockpit.
3. Alexander Schleicher TM 11 changes the Variant ASG 32 Mi into the Model ASG 32 Mi for which Section C of this TCDS applies





**Section B: ASG 32 EI**

**B.I General**

- |   |  |
|---|--|
| 1. Type/ Model/ Variant                     |  |
| 1.1 Type:                                   | ASG 32   |
| 1.2 Model:                                  | ASG 32 EI  |
| 2. Airworthiness Category                   | Sailplane, CS-22 - Utility                       |
| 3. Manufacturer                             | Alexander Schleicher GmbH & Co. Segelflugzeugbau |
| 4. EASA Type Certification Application Date | 28 October 2013                                  |
| 5. EASA Type Certification Date             | 22 December 2017                                 |

**B.II EASA Certification Basis**

- |   |   |
|---|---|
| 1. Reference Date for determining the applicable requirements |   |
| 2. Airworthiness Requirements                                 | Certification Specification for Sailplanes and Powered Sailplanes, issued 24. September 2008 (CS-22, Amdt. 2) |
| 3. Special Conditions   | CRI E-101 – Electrical Propulsion<br>CRI H-101 – Electrical Engine  |
| 4. Exemptions   | None  |
| 5. (Reserved) Deviations                                      | None  |
| 6. Equivalent Safety Findings                                 | CS 22.335 (f) calculation of $V_D$ according to OSTIV<br>CS 22.585(a) reduced by factor 1.2                   |
| 7. Environmental Protection                                   | ICAO Annex 16 (details refer to TCDSN EASA.A.599)   |



### **B.III Technical Characteristics and Operational Limitations**

1. Type Design Definition List of the drawing files ASG 32 EI, issue 15.12.2017
2. Description  
Double-seat, mid-wing CRP/GRP/ARP-composite construction for FAI 20m class with flaps, double-panel Schempp-Hirth airbrakes on upper wing surface, winglets, water ballast tanks in the wing and optional in the fuselage, retractable landing gear with hydraulic disc brake, T-shaped horizontal tail (fixed horizontal stabilizer with elevator) fin and rudder.  
  
Self-sustaining, electrical power-plant mounted in the centre fuselage.
3. Equipment  
Min. required Equipment:  
1 Air speed indicator (up to 300 km/h) (front seat)  
1 Altimeter (front seat)  
1 Magnetic compass (front seat)  
1 Power-plant instrument (front seat)  
1 Rear view mirror  
1 Outside air temperature gauge (front seat)  
1 4-Point safety harness (symmetrical) for each occupant  
1 Parachute or Cushion for back rest (~ 8cm thickness) for each occupant  
Additional Equipment refer to Flight and Maintenance Manual  
  
Additionally required for instruction or of the pilot in command sits in the rear seat:  
1 Air seed indicator in the rear seat (up to 300 km/h)  
1 Altimeter in the rear seat
4. Dimensions  
Span: 20,00 m  
Wing area: 15,70 m<sup>2</sup>  
Length: 9,07 m  
Height: 1,84 m
5. Engine
  - 5.1 Model Alexander Schleicher EA900/1-25LK
  - 5.2 Type Certificate n/a (accepted as part of the airframe)
  - 5.3 Limitations Maximum Power: 25 kW at 3000 rpm
  - 5.4 Max. continuous revs 2500 rpm
  - 5.5 Max. overspeed revs 3000 rpm
  - 5.6 Max. motor temperature 110°C
  - 5.7 Max. power electronics temperature 80°C



## 6. Propeller

6.1 Model	Alexander Schleicher AS2F1-4/L155-88-N3
6.2 Type Certificate	EASA.P.004
6.3 Number of blades	2
6.4 Diameter	1550 mm +3mm / -10 mm
6.5 Sense of Rotation	Left

## 7. Battery

7.1 Battery capacity	26 Ah
7.2 Non-usable battery capacity	10 Ah (39%)
7.3 Max battery discharge temperature	60°C
7.4 Min battery discharge temperature	-20°C
7.5 Max battery charge temperature	50°C
7.6 Min battery charge temperature	0°C
7.7 Range of permissible cell voltage	3 – 4,15 V

## 8. Launching Hooks

Nose tow hook "E 22", LBA Datasheet No. 11.402/9  
NTS  
Safety hook „Europa G 88“, LBA Datasheet No.  
60.230/2

## 9. Weak Links

Ultimate Strength:  
- for winch- and auto-tow launching max. 1100 daN  
- for aero-tow max. 1100 daN

## 10. Load Factors

+5,3 / -2,65 (up to  $V_A$ )  
+4,0 / -1,5 (up to  $V_{NE}$ )

## 11. Air Speeds

Manoeuvring Speed	$V_A$	180 km/h
Never Exceed Speed	$V_{NE}$	270 km/h
Maximum permitted Speeds		
- with flaps at 1, 2, 3, 4	$V_{FE}$	270 km/h
- with flaps at 5,6	$V_{FE}$	180 km/h
- with flaps at L	$V_{FE}$	150 km/h
- in rough air	$V_{RA}$	180 km/h
- for winch launching	$V_W$	140 km/h
- for aerotowing	$V_T$	180 km/h
- for gear operation	$V_{LO}$	180 km/h
- for propeller operation	$V_{PO}$	120 km/h
- with propeller extended	$V_{PE}$	180 km/h

## 12. Approved Operations Capability

VFR-Day  
Cloud flying not permitted  
Aerobatic manoeuvres are not permitted, except  
spinning

## 13. Launch methods

Aerotow  
Winch and Auto-Tow

## 14. Maximum Masses

Max. Mass: 850 kg  
Max. Mass of Non-Lifting Parts: 550 kg



15. Centre of Gravity Range	156 mm – 385 mm aft of datum
16. Datum	Wing leading edge at root rib
17. Levelling Means	Slope 1000 : 27 placed on upper side of fuselage boom horizontal
18. Control Surface Deflections	Refer to Maintenance Manual
19. Minimum Flight Crew	1
20. Maximum Passenger Seating Capacity	1
21. Baggage/ Cargo Compartments	9 kg
22. Lifetime limitations	Refer to Maintenance Manual



#### **B.IV Operating and Service Instructions**

1. Flight Manual ASG 32 EI, issue 15.09.2017, or later EASA approved revisions
2. Maintenance Manual ASG 32 EI, issue 01.04.2017, or later revisions
3. General Repair Manual for Alexander Schleicher Sailplanes and Powered Sialplanes, latest revision
4. Operating Manual and Maintenance Manual for Engine Alexander Schleicher EA900, latest approved version \*)
5. Operating Manual and Maintenance Manual for Propeller Alexander Schleicher AS2F1-4, latest approved version \*)
6. Manual for the TOST Release, latest approved version

\*) The operation and maintenance manuals are elements of the operation instructions of the ASG 32 EI. Necessary revisions are not be done in the manuals of the ASG 32 EI but separately by the engine and propeller manufacturer.



## **B.V** Notes

1. Manufacturing is confined to industrial production.
2. The surface colour of all fibre reinforced parts, which are exposed to sun radiation, must be painted either in
  - White
  - RAL 2004 (Reinorange)
  - RAL 2009 (Verkehrsorange)
  - RAL 3020 (Verkehrsröt)
  - or other colours listed in the maintenance manual section 13.4, maintenance instruction “coloured surfaces”

Exceptions are the areas for markings and registration, engine bay and cockpit.



**Section C: ASG 32 Mi**

**C.I General**

- |   |  |
|---|--|
| 1. Type/ Model/ Variant                     |  |
| 1.1 Type:                                   | ASG 32   |
| 1.2 Model:                                  | ASG 32 Mi  |
| 2. Airworthiness Category                   | Sailplane, CS-22 - Utility                       |
| 3. Manufacturer                             | Alexander Schleicher GmbH & Co. Segelflugzeugbau |
| 4. EASA Type Certification Application Date | 01 November 2012                                 |
| 5. EASA Type Certification Date             | 11 Februray 2016                                 |

**C.II EASA Certification Basis**

- |   |   |
|---|---|
| 1. Reference Date for determining the applicable requirements |   |
| 2. Airworthiness Requirements                                 | Certification Specification for Sailplanes and Powered Sailplanes, issued 24. September 2008 (CS-22, Amdt. 2) |
| 3. Special Conditions   | None  |
| 4. Exemptions   | None  |
| 5. (Reserved) Deviations                                      | None  |
| 6. Equivalent Safety Findings                                 | CS 22.335 (f) calculation of $V_D$ according to OSTIV<br>CS 22.585(a) reduced by factor 1.2                   |
| 7. Environmental Protection                                   | ICAO Annex 16 (details refer to TCDSN EASA.A.599)   |



### **C.III Technical Characteristics and Operational Limitations**

1. Type Design Definition
  2. Description
  3. Equipment
  4. Dimensions
  5. Engine
- List of the drawing files ASG 32 Mi, issue 31.01.2016
- Double-seat, self-launching powered sailplane, mid-wing CRP/GRP/ARP-composite construction for FAI 20m class with flaps, double-panel Schempp-Hirth airbrakes on upper wing surface, winglets, water ballast tanks in the wing and optional in the fuselage, retractable landing gear with hydraulic disc brake, T-shaped horizontal tail (fixed horizontal stabilizer with elevator) fin and rudder. Retractable power-plant mounted in the centre fuselage.
- Min. required Equipment:
- 1 Air speed indicator (up to 300 km/h) (front seat)
  - 1 Altimeter (front seat)
  - 1 Outside air temperature gauge (front seat)
  - 2 4-Point safety harness (symmetrical)
  - 2 Parachute or Cushion for back rest (~ 8cm thickness)
- Additional Equipment refer to Flight and Maintenance Manual
- With engine installed:
- 1 Magnetic compass (front seat)
  - 1 Power-plant instrument, type ILEC MCU ASH 30Mi (front seat)
  - 1 Rear view mirror (front seat)
- Span: 20,00 m  
Wing area: 15,70 m<sup>2</sup>  
Length: 9,07 m  
Height: 1,84 m
- 5.1 Model Austro Engine IAE50R-AA  
5.2 Type Certificate EASA.E.085  
5.3 Limitations Maximum Take-off Power (max. 3 min.): 37,3 kW at 7750 rpm  
5.4 Maximum Continuous Power 35,8 kW at 7100 rpm





## 6. Propeller

6.1 Model	Alexander Schleicher AS2F1-1/R153-92-N1
6.2 Type Certificate	EASA.P.004
6.3 Number of blades	2
6.4 Diameter	1530 mm ± 5 mm
6.5 Sense of Rotation or	Right
6.6 Model	Alexander Schleicher AS2F1-5/R153-88-N1
6.7 Type Certificate	EASA.P.004
6.8 Number of blades	2
6.9 Diameter	1530 mm ± 5 mm
6.10 Sense of Rotation	Right

## 7. Fuel capacities/Battery

7.1 Tank in the fuselage	14 l
7.2 Tank in right wing	15 l
7.3 Tank in left wing	15 l
7.4 Non-usable fuel	0,4 l

## 8. Launching Hooks

Nose tow hook "E 22", LBA Datasheet No. 11.402/9  
NTS  
Safety hook „Europa G 88“, LBA Datasheet No.  
60.230/2

## 9. Weak Links

Ultimate Strength:  
- for winch- and auto-tow launching max. 1100 daN  
- for aero-tow max. 1100 daN

## 10. Load Factors

+5,3 / -2,65 (up to  $V_A$ )  
+4,0 / -1,5 (up to  $V_{NE}$ )

## 11. Air Speeds

Manoeuvring Speed	$V_A$	180 km/h
Never Exceed Speed	$V_{NE}$	270 km/h
Maximum permitted Speeds		
- with flaps at 1, 2, 3, 4	$V_{FE}$	270 km/h
- with flaps at 5,6	$V_{FE}$	180 km/h
- with flaps at L	$V_{FE}$	150 km/h
- in rough air	$V_{RA}$	180 km/h
- for winch launching	$V_W$	140 km/h
- for aerotowing	$V_T$	180 km/h
- for gear operation	$V_{LO}$	180 km/h
- for propeller operation	$V_{PO}$	120 km/h
- with propeller extended	$V_{PE}$	180 km/h

## 12. Approved Operations Capability

VFR-Day  
Cloud flying not permitted  
Aerobatic manoeuvres not permitted, except Spinning

## 13. Launch methods

Aerotow  
Winch and Auto-Tow  
Self-Launch



14. Maximum Masses	Max. Mass: 850 kg Max. Mass of Non-Lifting Parts: 550 kg
15. Centre of Gravity Range	156 mm – 385 mm aft of datum
16. Datum	Wing leading edge at root rib
17. Levelling Means	Slope 1000 : 27 placed on upper side of fuselage boom horizontal
18. Control Surface Deflections	Refer to Maintenance Manual
19. Minimum Flight Crew	1
20. Maximum Passenger Seating Capacity	1
21. Baggage/ Cargo Compartments	9 kg
22. Lifetime limitations	Refer to Maintenance Manual



#### **C.IV Operating and Service Instructions**

1. Flight Manual Flight Manual ASG 32 Mi, issue 01.12.2015, or later EASA approved revisions
2. Maintenance Manual Maintenance Manual ASG 32 Mi, issue 15.01.2016, or later revisions
3. Structural Repair Manual General Repair Manual for Alexander Schleicher Sailplanes and Powered Sailplanes, latest revision
4. Operating Manual and Maintenance Manual for Engine Austro Engine IAE50R-AA series, latest approved version \*)
5. Operating Manual and Maintenance Manual for Propeller Alexander Schleicher AS2F1-1, latest approved version \*), or as applicable
6. Operating Manual and Maintenance Manual for Propeller Alexander Schleicher AS2F1-5, latest approved version \*)
7. Manual for the TOST Release, latest approved version

\*) The operation and maintenance manuals are elements of the operation instructions of the ASG 32 Mi. Necessary revisions are not be done in the manuals of the ASG 32 Mi but separately by the engine and propeller manufacturer.



## **C.V** Notes

1. Manufacturing is confined to industrial production.
2. The surface colour of all fibre reinforced parts, which are exposed to sun radiation, must be painted either in
  - White
  - RAL 2004 (Reinorange)
  - RAL 2009 (Verkehrsorange)
  - RAL 3020 (Verkehrsrrot)
  - or other colours listed in the maintenance manual section 13.4, maintenance instruction “coloured surfaces”Exceptions are the areas for markings and registration, engine bay and cockpit.
3. The model ASG 32 Mi has previously been addressed by Section A as variant ASG 32 Mi. The optional implementation of the AS Technical Note 11 converts the variant ASG 32 Mi into the model ASG 32 Mi.



**Section D: Administrative Section**

**D.I Acronyms & Abbreviations**

AS Alexander Schleicher GmbH & Co. Segelflugzeugbau  
VFR Visual Flight Rules  
MTOM Maximum Take-off Mass  
EASA European Union Aviation Safety Agency

**D.II Type Certificate Holder Record**

Alexander Schleicher GmbH & Co. Segelflugzeugbau  
Alexander-Schleicher-Str. 1  
36163 Poppenhausen  
Germany

**D.III Change Record**

Issue	Date	Changes	TC Issue No. & Date
Issue 01	11 February 2016	Initial Issue	Initial Issue, 11 February 2016
Issue 02	17 March 2016	Correction missing SC in A.II.3, separate manuals for both variants.	
Issue 03	11 January 2018	Addition of model ASG 32 EI	11 January 2018
Issue 04	06 June 2019	Alternative propeller for model ASG 32 Mi	
Issue 05	23 August 2021	Split of model ASG 32 with its variants ASG 32 and ASG 32 Mi into two models: ASG 32 and ASG 32 Mi See notes of Section A and C.	16 August 2021

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