EASA

TYPE-CERTIFICATE
DATA SHEET

EASA.A.024

L-13 “BLANÍK”

Type Certificate Holder:

Blaník Aircraft CZ s.r.o.
Beranových 65
199 00 Praha 9 - Letňany
Czech Republic

For models:
L-13 “BLANÍK”
L - 13 AC BLANÍK
L 13 A Blaník

Issue 10: 13 September 2019
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A.I. General (L - 13 "BLANÍK")

Data Sheet No.: EASA.A.024 Issue: 10 Date: 13 September 2019

1. Model: L - 13 "BLANÍK"

2. Airworthiness Category: Acrobatic Cloud flying

3. Type Certificate Holder: Blaník Aircraft CZ s.r.o. Beranových 65 190 00 Praha 9 - Letňany Czech Republic


5. Certification Application Date: ---

6. CAA CZ Type Certification Date: May 29, 1959

7. EASA Type Certification Date: February 4, 2005

8. The EASA Type Certificate replaces Czech Type Certificate No. 2725-59
A.II. Certification Basis

1. Reference Date for determining the applicable requirements: ---

2. Certification Basis:
   - Bauvorschriften für Segelflugzeuge (BSV) issued August 1939
   - BCAR, Section E, issued June 16, 1966

3. Airworthiness Requirements: see (2) above

4. Requirements elected to comply:
   - CS 22.627, Amdt. 2 (Fatigue strength) for sailplanes with installed modification as per Mandatory Bulletin L13/117a initial issue (covering Major Change TDC-002-L13-BL) or its further approved revisions.

5. EASA Special Conditions: None

6. EASA Exemptions: None

7. EASA Equivalent Safety Findings: None

A.III. Technical Characteristics and Operational Limitations

1. Type Design Definition: Drawing No. A 101 310 N

2. Description:
   - All-metal, cantilever, high-wing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake, and a tail skid or (an optional) tail wheel. Horizontal tail surfaces consist of a two-piece tail-plane and elevator, vertical tail surfaces consist of a fin and a rudder.

3. Equipment:
   - Minimum equipment:
     - 2 airspeed indicators, range to 400 km/h
     - 2 altimeters
     - 2 four-point safety harnesses (symmetric)
     - 2 parachutes or backrests (approx. 10 cm thick when compressed)

4. Dimensions:
   - Span: 16.2 m
   - Length: 8.4 m
   - Height: 2.1 m
   - Wing Area: 19.15 m²
   - Aspect Ratio: 13.7

5. Launching Hooks:
   - Nose tow release Dwg. No. A 740 210 N, or nose tow release "E85", LBA Type Certificate No. 60.230/1
   - Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P
   - "Europa G 72" safety tow release LBA Type Certificate No. 60.230/2, or
   - "Europa G 73" safety tow release LBA Type Certificate No. 60.230/2, or
   - "Europa G 88" safety tow release LBA Type Certificate No. 60.230/2
6. Weak links:

- for winch launching max. 6300 N
- for aero-tow max. 6300 N

7. Air Speeds:

- Manoeuvering Speed $V_A$: 145 km/h IAS
- Never Exceed Speed $V_{NE}$:
  - up to 2500 m MSL: 253 km/h IAS
  - 2500 to 3000 m MSL: 245 km/h IAS
  - 3000 to 4000 m MSL: 230 km/h IAS
  - 4000 to 5000 m MSL: 215 km/h IAS
  - 5000 to 6000 m MSL: 201 km/h IAS
  - 6000 to 7000 m MSL: 187 km/h IAS
  - 7000 to 8000 m MSL: 174 km/h IAS
  - 8000 to 9000 m MSL: 161 km/h IAS
  - 9000 to 10000 m MSL: 150 km/h IAS
- Rough Air Speed $V_{RA}$: 145 km/h IAS
- Max. Aerotow Speed $V_T$: 140 km/h IAS
- Max. Flap Extended Speed $V_{FE}$: 110 km/h IAS
- Max. Winch-launch Speed $V_W$: 120 km/h IAS
- Max. Landing Gear Operating Speed $V_{LO}$: ---

8. Operational Capability:

- VFR Day
- Cloud flying

9. Maximum Weights:

- Maximum Weight: 500 kg
- Maximum Weight of non-lifting parts: 340 kg
- Empty Weight: 292 kg ± 2%

10. Centre of Gravity Range:

- Fore most c.g. limit aft of reference plane 112 mm
- Aft most c.g. limit aft of reference plane 300 mm
  [MAC is 1253 mm]

11. Datum:

- Wing leading edge at root rib

12. Levelling Means:

- Leveling points on fuselage

13. Minimum Flight Crew:

- 1 (Pilot)

14. Maximum Passenger Seating Capacity:

- 1

15. Lifetime limitations:

- Refer to Maintenance Manual

16. Deflection angles of control surfaces:

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<thead>
<tr>
<th>Control Surface</th>
<th>Up</th>
<th>Down</th>
</tr>
</thead>
<tbody>
<tr>
<td>Elevator</td>
<td>$32° + 2°$</td>
<td>$25° + 1°$</td>
</tr>
<tr>
<td>Rudder</td>
<td>$30° + 1°$</td>
<td></td>
</tr>
<tr>
<td>Ailerons</td>
<td>$34° + 2°$</td>
<td>$13° + 2°$</td>
</tr>
<tr>
<td>Wing flaps</td>
<td>$08° + 1°$</td>
<td></td>
</tr>
<tr>
<td>Elevator trim tab</td>
<td>$12° + 1°$</td>
<td>$35° + 1°$</td>
</tr>
</tbody>
</table>
A.IV. Operating and Service Instructions

1. Flight Manual:

   - **In Czech language**
     Do-L13-1111.1
     Směrnice pro pilota větroně L 13
   - **In English language**
     Do-L13-1111.3
     Pilots Notes for the L-13 Sailplane
   - **In German language**
     Do-L13-1111.2
     Fluganweisung fur das Segelflugzeug L-13
   - **In Russian language**
     Do-L13-1111.5
     Rukovodstvo po letnoj ekspluataci planera L 13
   - **In Spanish language**
     Do-L13-1111.4
     Planeador L 13 Blaník Instructiones Para el Piloto
   - **In English language**
     Do-L13-1111.6
     L-13 „BLANÍK“ Sailplane Flight Manual *)

*) For sailplanes with installed modification as per Mandatory Bulletin L13/117a initial issue (covering Major Change TDC-002-L13-BL) or its further approved revisions.

2. Maintenance Manual:

   - **In Czech language**
     Do-L13-1132.1
     Technická příručka větroně L 13
     Do-L13-1131.1
     Příručka pro provoz a údržbu větroně L 13 Blaník bez generálních oprav
   - **In English language**
     Do-L13-1132.3
     Technical Manual of the L 13 Sailplane,
     Do-L13-1131.3
     Manual for Operation and Maintenance of the L 13 Blaník Sailplane without overhauls
   - **In German language**
     Do-L13-1132.2
     Technisches Handbuch fur das Segelflugzeug L-13
   - **In Russian language**
     Do-L13-1131.5
     Rukovodstvo po techničeskoj ekspluatacii L 13 Blaník bez kapitalnych remontov

3. Illustrated Parts Catalogue:

   - **In Czech language**
     Do-L13-2121.6
     Kusovník větroně L 13 Blaník (C-A-N)
   - **In English language**
     Do-L13-2121.6
     Spare Parts Catalogue L 13 Blaník (C-A-N)
   - **In German language**
     Do-L13 2121.6
     Katalog der Bestandteile L-13 Blaník

4. Overhaul Manual

   - **In Czech language**
     Do-L13-3031.1
     Příručka pro generální opravu kluzáku L 13, L 13A
   - **In English language**
     Do-L13-3031.3
     Overhaul Manual for L 13, L 13A Gliders
   - **In Russian language**
     Do-L13-3131.5
     Rukovodstvo po kapitalnomu remontu planera L 13

5. Operating Manuals for Tow Releases

   - **In German and in English language**
     Operating Manual for Nose Tow Releases TOST “Europa E 85”
     Operating Manual for Safety Tow Releases TOST “Europa G 88”
     Operating Manual for Safety Tow Releases TOST “Europa G 72” and “Europa G 73”
A.V. Notes

1. Six-position Serial Numbers starting with 17 precede the six-position Serial Numbers starting with 02.
2. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly theContracted DOA Holder.
3. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.
4. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.
### B.I. General (L - 13 AC BLANÍK)

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<th>Issue: 10 Date: 13 September 2019</th>
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<td>2. Airworthiness Category: Acrobatic Cloud flying</td>
<td></td>
</tr>
<tr>
<td>3. Type Certificate Holder: Blanik Aircraft CZ s.r.o. Beranových 65 190 00 Praha 9 - Letňany Czech Republic</td>
<td></td>
</tr>
<tr>
<td>5. Certification Application Date: ---</td>
<td></td>
</tr>
<tr>
<td>6. CAA CZ Type Certification Date: July 15, 1999</td>
<td></td>
</tr>
<tr>
<td>7. EASA Type Certification Date: February 4, 2005</td>
<td></td>
</tr>
<tr>
<td>8. The EASA Type Certificate replaces Czech Type Certificate No. 2725-59</td>
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</tr>
</tbody>
</table>

### B.II. Certification Basis

| 1. Reference Date for determining the applicable requirements: August 31, 1998 |
| 3. Airworthiness Requirements: BCAR, Section E, issued June 6, 1966 |
| 4. Requirements elected to comply: None |
| 5. EASA Special Conditions: Appendix H, Joint Airworthiness Requirements, Sailplanes and Powered Sailplanes, Change 5 of October 28, 1995 |
| 6. EASA Exemptions: None |
| 7. EASA Equivalent Safety Findings: None |
B.III. Technical Characteristics and Operational Limitations

1. Type Design Definition: Drawing No. A 500 020 N

2. Description: L-13 AC BLANÍK sailplane is all-metal, cantilever, highwing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake and a tail wheel. Horizontal tail surfaces consist of a two-piece tail-plane and elevator, vertical tail surfaces consist of a fin and a rudder. The S/N 018901 and from S/N 039102 there are installed ailerons outer stops, left aileron balance tab and there exists an option of wing tip extensions installation. Among the standard equipment there belongs AMU-1B recording unit.

3. Equipment: Minimum equipment:
   - 2 airspeed indicators, range to 400 km/h
   - 2 altimeters
   - 2 five-point safety harnesses (symmetric)
   - 2 accelerometers
   - 2 parachutes or backrests (approx. 10 cm thick when compressed)
   - 1 AMU-1B recording unit

4. Dimensions:
   - Span
     - 14.2 m without wing-tip extensions
     - 16.2 m with wing-tip extensions
   - Length
     - 8.4 m
   - Height
     - 2.09 m
   - Wing Area
     - 17.44 m² without wing-tip extensions
     - 19.15 m² with wing-tip extensions
   - Aspect Ratio
     - 11.186 without wing-tip extensions
     - 13.7 with wing-tip extensions

5. Launching Hooks: Nose tow release Dwg. No. A 740 210 N, or nose tow release "E85", LBA Type Certificate No. 60.230/1
   Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P
   "Europa G 88" safety tow release LBA Type Certificate No. 60.230/2

6. Weak links: Ultimate Strength:
   - for winch launching max. 6230 N
   - for aero-tow max. 6230 N

7. Air Speeds: Air speeds of sailplane without wing tip extensions:
   - Manoeuvring Speed $V_A$ 160 km/h IAS
   - Never Exceed Speed $V_{NE}$
     - up to 2500 m MSL 230 km/h IAS
     - 2500 to 3000 m MSL 223 km/h IAS
     - 3000 to 4000 m MSL 209 km/h IAS
     - 4000 to 5000 m MSL 195 km/h IAS
     - 5000 to 6000 m MSL 182 km/h IAS
     - 6000 to 7000 m MSL 170 km/h IAS
     - 7000 to 8000 m MSL 158 km/h IAS
     - 8000 to 9000 m MSL 147 km/h IAS
9000 to 10000 m MSL 136 km/h IAS
Rough Air Speed $V_{RA}$ 160 km/h
Max. Aerotow Speed $V_T$ 150 km/h
Max. Winch-launch Speed $V_W$ 120 km/h
Max. Landing Gear Operating Speed $V_{LO}$ ---

Maneuvering Speed $V_A$ 150 km/h IAS
Never Exceed Speed $V_{NE}$

\begin{align*}
\text{up to 2500 m MSL} & \quad 230 \text{ km/h IAS} \\
2500 \text{ to 3000 m MSL} & \quad 223 \text{ km/h IAS} \\
3000 \text{ to 4000 m MSL} & \quad 209 \text{ km/h IAS} \\
4000 \text{ to 5000 m MSL} & \quad 195 \text{ km/h IAS} \\
5000 \text{ to 6000 m MSL} & \quad 182 \text{ km/h IAS} \\
6000 \text{ to 7000 m MSL} & \quad 170 \text{ km/h IAS} \\
7000 \text{ to 8000 m MSL} & \quad 158 \text{ km/h IAS} \\
8000 \text{ to 9000 m MSL} & \quad 147 \text{ km/h IAS} \\
9000 \text{ to 10000 m MSL} & \quad 136 \text{ km/h IAS} \\
\end{align*}
Rough Air Speed $V_{RA}$ 150 km/h
Max. Aerotow Speed $V_T$ 150 km/h
Max. Winch-launch Speed $V_W$ 120 km/h
Max. Landing Gear Operating Speed $V_{LO}$ ---

8. Operational Capability: VFR Day
Cloud flying

9. Maximum Weights:
Maximum Weight: 500 kg (without extensions)
510 kg (with extensions)
Maximum Weight of non-lifting parts: 355 kg
Empty Weight: 305 kg ± 2% (without extensions)
315 kg ± 2% (with extensions)

10. Centre of Gravity Range:
Fore most c.g. limit aft of reference plane 143 mm
Aft most c.g. limit aft of reference plane 337 mm

MAC is 1295 mm without wing tip extensions
MAC is 1252.3 mm with wing tip extensions

11. Datum:
Wing leading edge at root rib

12. Levelling Means:
Leveling points on fuselage in horizontal position.


14. Maximum Passenger Seating Capacity: 1

15. Lifetime limitations: Refer to Maintenance Manual

16. Deflection angles of control surfaces:
\begin{align*}
\text{Elevator} & \quad \text{up} \quad 32° ± 2° \\
& \quad \text{down} \quad 27° ± 1° \\
\text{Rudder} & \quad \text{left, right} \quad 29° ± 1° \\
\text{Ailerons} & \quad \text{up} \quad 34° + 2° \\
& \quad \text{down} \quad 13° + 2° \\
\text{Left aileron balance tab} & \quad \text{up} \quad 20° ± 2° \\
& \quad \text{down} \quad 15° ± 2° \\
\text{Elevator trim tab} & \quad \text{up} \quad 12° ± 1° \\
& \quad \text{down} \quad 35° ± 1°
\end{align*}
B.IV. Operating and Service Instructions

1. Flight Manual:
   - In Czech language
     Do-L13AC-1013.1 Letová příručka L 13 AC Blaník (to S/N 008606 from S/N 028902 to 029101)
     Do-L13AC-1014.0 Letová příručka L 13 AC Blaník (S/N 018901 from S/N 039102)
   - In English language
     Do-L13AC-1013.3 Sailplane Flight Manual L 13AC Blaník to S/N 008606 from S/N 028902 to 029101
     Do-L13AC-1014.2 Sailplane Flight Manual L 13AC Blaník (S/N 018901 from S/N039102)

2. Maintenance Manual:
   - In Czech language
     Do-L13AC-1032.1 Provozně technická příručka kluzáku L 13 AC
   - In English language
     Do-L13AC-1032.3 Maintenance Manual for the Sailplane L 13 AC

3. Illustrated Parts Catalogue:
   - In Czech language
     Do-L13AC-2051.0 Katalog dílů a montážních jednotek kluzáku L-13AC Blaník (C/A)
   - In English language
     Do-L13AC-2051.0 Illustrated parts catalogue for the sailplane L 13 AC Blaník (C/A)

4. Operation book
   - In Czech language
     Do-L13AC.1015.02 L 13AC Blaník Záznamník provozu kluzáku (C/A)
   - In English language
     Do-L13AC.1015.02 L 13 AC Blaník Sailplane operation book of records (C/A)

5. Operating Manuals for Tow Releases
   - In German and in English language
     Operating Manual for Nose Tow Releases TOST "Europa E 85"
     Operating Manual for Safety Tow Releases TOST "Europa G 88"

B.V. Notes

1. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.

2. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.

3. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.
C.I. General  
(L 13 A Blaník)

Data Sheet No.: EASA.A.024  
Issue: 10  
Date: 13 September 2019

1. Model:  
L 13 A Blaník

2. Airworthiness Category:  
Acrobatic  
Cloud flying

3. Type Certificate Holder:  
Blaník Aircraft CZ s.r.o.  
Beranových 65  
190 00 Praha 9 - Letňany  
Czech Republic

4. Manufacturer:  
from S/N 817401 to S/N 827420  
LET, n.p.  
686 04 Kunovice 1177  
CZECH REPUBLIC  

from S/N 968501 to S/N 968505  
and S/N 827421  
LET, a.s.  
686 04 Kunovice 1177  
CZECH REPUBLIC

5. Certification Application Date:  
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6. CAA CZ Type Certification Date:  
December 16, 1981

7. EASA Type Certification Date:  
12 August 2005

8. The EASA Type Certificate replaces Czech Type Certificate No. 2725-59

C.II. Certification Basis

1. Reference Date for determining the applicable requirements:  
---

2. Certification Basis:  
BCAR, Section E, issued June 6, 1966

3. Airworthiness Requirements:  
see (2) above

4. Requirements elected to comply:  
None

5. EASA Special Conditions:  
None

6. EASA Exemptions:  
None

7. EASA Equivalent Safety Findings:  
None
C.III. Technical Characteristics and Operational Limitations

1. Type Design Definition:  Drawing No. A 101 310 N

2. Description:  All-metal, cantilever, high-wing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake, and a tail skid or (an optional) tail wheel. Horizontal tail surfaces consist of a two-piece tail-plane and elevator, vertical tail surfaces consist of a fin and a rudder.

3. Equipment:  Minimum equipment:
   - 2 airspeed indicators, range to 400 km/h
   - 2 altimeters
   - 2 four-point safety harnesses (symmetric)
   - 2 parachutes or backrests (approx. 10 cm thick when compressed)

4. Dimensions:
   - Span: 16.2 m
   - Length: 8.4 m
   - Height: 2.1 m
   - Wing Area: 19.15 m²
   - Aspect Ratio: 13.7

5. Launching Hooks:  Nose tow release of type SR-L13.225, or nose tow release Dwg. No. A 740 210 N, or nose tow release "E85", LBA Type Certificate No. 60.230/1
   - Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P
   - "Europa G 72" safety tow release LBA Type Certificate No. 60.230/2, or
   - "Europa G 73" safety tow release LBA Type Certificate No. 60.230/2, or
   - "Europa G 88" safety tow release LBA Type Certificate No. 60.230/2

6. Weak links:  Ultimate Strength:
   - for winch launching  max. 6230 N
   - for aero-tow  max. 6230 N

7. Air Speeds:
   - Manoeuvering Speed \( V_A \): 145 km/h IAS
   - Never Exceed Speed \( V_{NE} \):
     - up to 2500 m MSL  253 km/h IAS
     - 2500 to 3000 m MSL  245 km/h IAS
     - 3000 to 4000 m MSL  230 km/h IAS
     - 4000 to 5000 m MSL  215 km/h IAS
     - 5000 to 6000 m MSL  201 km/h IAS
     - 6000 to 7000 m MSL  187 km/h IAS
     - 7000 to 8000 m MSL  174 km/h IAS
     - 8000 to 9000 m MSL  161 km/h IAS
     - 9000 to 10000 m MSL  150 km/h IAS
   - Rough Air Speed \( V_{RA} \): 145 km/h
   - Max. Aerotow Speed \( V_T \): 140 km/h
   - Max. Winch-launch Speed \( V_W \): 120 km/h
   - Max. Landing Gear Operating Speed \( V_{LO} \): ---
8. Operational Capability: VFR Day
   Cloud flying

9. Maximum Weights:
   Maximum Weight: 500 kg
   Maximum Weight of non-lifting parts: 340 kg
   Empty Weight: 306 kg ± 2%

10. Centre of Gravity Range:
    Fore most c.g. limit aft of reference plane 112 mm
    Aft most c.g. limit aft of reference plane 300 mm
    [MAC is 1253 mm]

11. Datum:
    Wing leading edge at root rib

12. Levelling Means:
    Leveling points on fuselage in horizontal position.

13. Minimum Flight Crew:
    1 (Pilot)

14. Maximum Passenger Seating Capacity:
    1

15. Lifetime limitations:
    Refer to Maintenance Manual

16. Deflection angles of control surfaces:
    Elevator       up   32° + 2°
                   down  25° ± 1°
    Rudder        left, right  30° + 1°
    Ailerons      up   34° + 2°
                   down  13° ± 2°
    Wing flaps      down  8° ± 1°
    Elevator trim tab
        up      12° ± 1°
        down   35° ± 1°

C.IV. Operating and Service Instructions

1. Flight Manual:
   In Czech language
   Do-L13A-1011.1       Letová příručka L 13 A
   In English language
   Do-L13A-1011.3       Flight Manual of the L 13 A
   In German language
   Do-L13A-1011.2       Flughandbuch fur das Segelflug L 13 A

2. Maintenance Manual:
   In Czech language
   Do-L13A-1031.1       Příručka pro obsluhu, údržbu a opravy kluzáku L 13
   In English language
   Do-L13A-1031.3       Technical Manual of the L 13 A Sailplane
   In German language
   Do-L13A-1031.2       Handbuch fur die Reparatur, Instandhaltung und Wartung
def des Segelflugzeuges L 13 A
3. Illustrated Parts Catalogue:

In Czech language
Do-L13-2126.6  Kusovník větroně L 13 Blaník (C-A-N)

In English language
Do-L13-2126.6  Spare Parts Catalogue L 13 Blaník (C-A-N)

In German language
Do-L13-2126.6  Katalog der Bestandteile L-13 Blaník (C-A-N)

4. Overhaul Manual

In Czech language
Do-L13-3031.1  Příručka pro generální opravu kluzáku L 13, L 13A

In English language
Do-L13-3031.3  Overhaul Manual for L 13, L 13A Gliders

5. Sailplane Technical Description L 13A

In Czech language
Do-L13A-1021.1  Sailplane Technical Description L 13A

6. Operating Manuals for Tow Releases

In German and in English language
Operating Manual for Nose Tow Releases TOST "Europa E 85"
Operating Manual for Safety Tow Releases TOST "Europa G 88"
Operating Manual for Safety Tow Releases TOST "Europa G 72" and "Europa G 73"

C.V. Notes

1. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.

2. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.

3. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.
ADMINISTRATIVE SECTION

I. Acronyms

N/A

II. Type Certificate Holder Record

Up to 19 June 2013

Aircraft Industries, a.s.
Na Záhonech 1177
686 04 Kunovice
Czech Republic

Up to 09 January 2017

BLANIK LIMITED
2nd Floor Beaux Lane House
Mercer Street Lower
Dublin 2
Republic of Ireland

Up to 12 September 2019

Blanik Aircraft CZ s.r.o.
Karolinská 661/4
186 00 Praha 8
Czech Republic

Since 13 September 2019

Blanik Aircraft CZ s.r.o.
Beranových 65
190 00 Praha 9 - Letňany
Czech Republic
### III. Change Record

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<td>01</td>
<td>04 February 2005</td>
<td>Initial issue of TCDS No. EASA.A.044</td>
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<tr>
<td>02</td>
<td>12 August 2005</td>
<td>To record the change in the name of the TC holder and to include the L 13 A model.</td>
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<tr>
<td>03</td>
<td>09 September 2005</td>
<td>Editorial changes</td>
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<td>04</td>
<td>23 May 2006</td>
<td>Change in address of the TC holder</td>
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<td>05</td>
<td>20 June 2013</td>
<td>Change of the TC holder and new layout of TCDS</td>
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<tr>
<td>06</td>
<td>09 May 2016</td>
<td>Addition of S/N 827421 (L 13 A Blanik) to section C.I.5.</td>
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<tr>
<td>07</td>
<td>21 September 2016</td>
<td>Addition of CS 22.627 Eltect to Comply to section A.II.4. and addition of Sailplane Flight Manual (Do-L13-1111.6) to section A.IV.1.</td>
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<td>08</td>
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<td>10</td>
<td>13 September 2019</td>
<td>Change of the TC holder’s address</td>
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