



European Union Aviation Safety Agency

EASA

**TYPE-CERTIFICATE
DATA SHEET**

EASA.A.024

L-13 "BLANÍK"

Type Certificate Holder:

Blanik Aircraft CZ s.r.o.
Beranových 65
199 00 Praha 9 - Letňany
Czech Republic

For models: L-13 "BLANÍK"
L - 13 AC BLANÍK
L 13 A Blaník

Issue 10: 13 September 2019

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A.I. General (L - 13 "BLANÍK")

- Data Sheet No.: EASA.A.024 Issue: 10 Date: 13 September 2019
1. Model: L - 13 "BLANÍK"
 2. Airworthiness Category: Acrobatic
Cloud flying
 3. Type Certificate Holder: Blanik Aircraft CZ s.r.o.
Beranových 65
190 00 Praha 9 - Letňany
Czech Republic
 4. Manufacturer:

from S/N 170101 to S/N 173337,
from S/N 173341 to S/N 173343,
from S/N 173345 to S/N 173346,
S/N 173350,
S/N 173353,
from S/N 173356 to S/N 173359,
from S/N 173402 to S/N 173404,
S/N 173407,
from S/N 173415 to S/N 173416,
from S/N 173422 to S/N 173439,
S/N 173441
Strojirny první pětiletky n.p.
Uherské Hradiště - Kunovice 1177
Czechoslovakia

from S/N 173338 to S/N 173340
S/N 173344
from S/N 173347 to S/N 173349
from S/N 173351 to S/N 173352
from S/N 173354 to S/N 173355
from S/N 173360 to S/N 173401
from S/N 173405 to S/N 173406
from S/N 173408 to S/N 173414
from S/N 173417 to S/N 173421
S/N 173440
from S/N 173442 to S/N 027361
LET, n.p.
Uherské Hradiště - Kunovice 1177
Czechoslovakia
 5. Certification Application Date: ---
 6. CAA CZ Type Certification Date: May 29, 1959
 7. EASA Type Certification Date: February 4, 2005
 8. The EASA Type Certificate replaces Czech Type Certificate No. 2725-59

A.II. Certification Basis

1. Reference Date for determining the applicable requirements: ---
2. Certification Basis:
 - Bauvorschriften für Segelflugzeuge (BSV) issued August 1939
 - BCAR, Section E, issued June 16, 1966
3. Airworthiness Requirements: see (2) above
4. Requirements elected to comply: CS 22.627, Amdt. 2 (Fatigue strength) for sailplanes with installed modification as per Mandatory Bulletin L13/117a initial issue (covering Major Change TDC-002-L13-BL) or its further approved revisions.
5. EASA Special Conditions: None
6. EASA Exemptions: None
7. EASA Equivalent Safety Findings: None

A.III. Technical Characteristics and Operational Limitations

1. Type Design Definition: Drawing No. A 101 310 N
2. Description: All-metal, cantilever, high-wing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake, and a tail skid or (an optional) tail wheel. Horizontal tail surfaces consist of a two-piece tail-plane and elevator, vertical tail surfaces consist of a fin and a rudder.
3. Equipment:
 - Minimum equipment:
 - 2 airspeed indicators, range to 400 km/h
 - 2 altimeters
 - 2 four-point safety harnesses (symmetric)
 - 2 parachutes or backrests (approx. 10 cm thick when compressed)
4. Dimensions:

| | |
|--------------|----------------------|
| Span | 16.2 m |
| Length | 8.4 m |
| Height | 2.1 m |
| Wing Area | 19.15 m ² |
| Aspect Ratio | 13,7 |
5. Launching Hooks:
 - Nose tow release Dwg. No. A 740 210 N, or nose tow release "E85", LBA Type Certificate No. 60.230/1
 - Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P
 - "Europa G 72" safety tow release LBA Type Certificate No. 60.230/2, or
 - "Europa G 73" safety tow release LBA Type Certificate No. 60.230/2, or
 - "Europa G 88" safety tow release LBA Type Certificate No. 60.230/2

- | | | |
|--|---|----------------------------|
| 6. Weak links: | Ultimate Strength: - for winch launching - for aero-tow | max. 6300 N max. 6300 N |
| 7. Air Speeds: | | |
| Manoeuvring Speed V_A | 145 km/h IAS | |
| Never Exceed Speed V_{NE} | | |
| up to 2500 m MSL | 253 km/h IAS | |
| 2500 to 3000 m MSL | 245 km/h IAS | |
| 3000 to 4000 m MSL | 230 km/h IAS | |
| 4000 to 5000 m MSL | 215 km/h IAS | |
| 5000 to 6000 m MSL | 201 km/h IAS | |
| 6000 to 7000 m MSL | 187 km/h IAS | |
| 7000 to 8000 m MSL | 174 km/h IAS | |
| 8000 to 9000 m MSL | 161 km/h IAS | |
| 9000 to 10000 m MSL | 150 km/h IAS | |
| Rough Air Speed V_{RA} | 145 km/h IAS | |
| Max. Aerotow Speed V_T | 140 km/h IAS | |
| Max. Flap Extended Speed V_{FE} | 110 km/h IAS | |
| Max. Winch-launch Speed V_W | 120 km/h IAS | |
| Max. Landing Gear Operating Speed V_{LO} | --- | |
| 8. Operational Capability: | VFR Day Cloud flying | |
| 9. Maximum Weights: | | |
| Maximum Weight: | 500 kg | |
| Maximum Weight of non-lifting parts: | 340 kg | |
| Empty Weight: | 292 kg \pm 2% | |
| 10. Centre of Gravity Range: | Fore most c.g. limit aft of reference plane 112 mm Aft most c.g. limit aft of reference plane 300 mm [MAC is 1253 mm] | |
| 11. Datum: | Wing leading edge at root rib | |
| 12. Levelling Means: | Leveling points on fuselage | |
| 13. Minimum Flight Crew: | 1 (Pilot) | |
| 14. Maximum Passenger Seating Capacity: | 1 | |
| 15. Lifetime limitations: | Refer to Maintenance Manual | |
| 16. Deflection angles of control surfaces: | | |
| Elevator | up | 32° + 2° |
| | down | 25° \pm 1° |
| Rudder | left, right | 30° + 1° |
| Ailerons | up | 34° + 2° |
| | down | 13° + 2° |
| Wing flaps | down | 08° \pm 1° |
| Elevator trim tab | up | 12° \pm 1° |
| | down | 35° \pm 1° |

A.IV. Operating and Service Instructions

1. Flight Manual:

| | |
|--------------------------------------|--|
| In Czech language Do-L13-1111.1 | Směrnice pro pilota větroně L 13 |
| In English language Do-L13-1111.3 | Pilots Notes for the L-13 Sailplane |
| In German language Do-L13-1111.2 | Fluganweisung für das Segelflugzeug L-13 |
| In Russian language Do-L13-1111.5 | Rukovodstvo po letnoj ekspluatácii planera L 13 |
| In Spanish language Do-L13-1111.4 | Planeador L 13 Blaník Instrucciones Para el Piloto |
| In English language Do-L13-1111.6 | L-13 „BLANÍK“ Sailplane Flight Manual *) |

*) For sailplanes with installed modification as per Mandatory Bulletin L13/117a initial issue (covering Major Change TDC-002-L13-BL) or its further approved revisions.

2. Maintenance Manual:

| | |
|---|--|
| In Czech language Do-L13-1132.1 Do-L13-1131.1 | Technická příručka větroně L 13 Příručka pro provoz a údržbu větroně L 13 Blaník bez generálních oprav |
| In English language Do-L13-1132.3 Do-L13-1131.3 | Technical Manual of the L 13 Sailplane, Manual for Operation and Maintenance of the L 13 Blaník Sailplane without overhauls |
| In German language Do-L13-1132.2 In Russian language Do-L13-1131.5 | Technisches Handbuch für das Segelflugzeug L-13 Rukovodstvo po tehničeskoj ekspluatácii L 13 Blaník bez kapitalnych remontov |

3. Illustrated Parts Catalogue:

| | |
|--------------------------------------|---|
| In Czech language Do L13-2121.6 | Kusovník větroně L 13 Blaník (C-A-N) |
| In English language Do-L13-2121.6 | Spare Parts Catalogue L 13 Blaník (C-A-N) |
| In German language Do-L13 2121.6 | Katalog der Bestandteile L-13 Blaník |

4. Overhaul Manual

| | |
|--------------------------------------|---|
| In Czech language Do-L13-3031.1 | Příručka pro generální opravu kluzáku L 13, L 13A |
| In English language Do-L13-3031.3 | Overhaul Manual for L 13, L 13A Gliders |
| In Russian language Do-L13-3131.5 | Rukovodstvo po kapitalnomu remontu planera L 13 |

5. Operating Manuals for Tow Releases

| | |
|-----------------------------------|---|
| In German and in English language | Operating Manual for Nose Tow Releases TOST "Europa E 85" |
| | Operating Manual for Safety Tow Releases TOST "Europa G 88" |
| | Operating Manual for Safety Tow Releases TOST "Europa G 72" and "Europa G 73" |

A.V. Notes

1. Six-position Serial Numbers starting with 17 precede the six-position Serial Numbers starting with 02.
2. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.
3. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.
4. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.

B.I. General (L - 13 AC BLANÍK)

| | | |
|--|--|-------------------------|
| Data Sheet No.: EASA.A.024 | Issue: 10 | Date: 13 September 2019 |
| 1. Model: | L - 13 AC BLANÍK | |
| 2. Airworthiness Category: | Acrobatic Cloud flying | |
| 3. Type Certificate Holder: | Blanik Aircraft CZ s.r.o. Beranových 65 190 00 Praha 9 - Letňany Czech Republic | |
| 4. Manufacturer: | from S/N 988601 to S/N 008606 LET, a.s. 686 04 Kunovice 1177 CZECH REPUBLIC S/N 018901 and from S/N 028902 to S/N 029101 LETECKÉ ZÁVODY a.s. 686 04 Kunovice 1177 CZECH REPUBLIC | |
| 5. Certification Application Date: | --- | |
| 6. CAA CZ Type Certification Date: | July 15, 1999 | |
| 7. EASA Type Certification Date: | February 4, 2005 | |
| 8. The EASA Type Certificate replaces Czech Type Certificate No. 2725-59 | | |

B.II. Certification Basis

| | |
|--|---|
| 1. Reference Date for determining the applicable requirements: | August 31, 1998 |
| 2. Certification Basis: | CRI-A-01, issue 2, issued August 31, 1998 |
| 3. Airworthiness Requirements: | BCAR, Section E, issued June 6, 1966 |
| 4. Requirements elected to comply: | None |
| 5. EASA Special Conditions: | Appendix H, Joint Airworthiness Requirements, Sailplanes and Powered Sailplanes, Change 5 of October 28, 1995 |
| 6. EASA Exemptions: | None |
| 7. EASA Equivalent Safety Findings: | None |

B.III. Technical Characteristics and Operational Limitations

1. Type Design Definition: Drawing No. A 500 020 N
2. Description: L - 13 AC BLANÍK sailplane is all-metal, cantilever, highwing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake and a tail wheel. Horizontal tail surfaces consist of a two-piece tail-plane and elevator, vertical tail surfaces consist of a fin and a rudder. The S/N 018901 and from S/N 039102 there are installed ailerons outer stops, left aileron balance tab and there exists an option of wing tip extensions installation. Among the standard equipment there belongs AMU-1B recording unit.
3. Equipment: Minimum equipment:
2 airspeed indicators, range to 400 km/h
2 altimeters
2 five-point safety harnesses (symmetric)
2 accelerometers
2 parachutes or backrests (approx. 10 cm thick when compressed)
1 AMU-1B recording unit
4. Dimensions:
Span 14.2 m without wing-tip extensions
16.2 m with wing-tip extensions
Length 8.4 m
Height 2.09 m
Wing Area 17.44 m² without wing-tip extensions
19.15 m² with wing-tip extensions
Aspect Ratio 11.186 without wing-tip extensions
13.7 with wing-tip extensions
5. Launching Hooks: Nose tow release Dwg. No. A 740 210 N, or nose tow release "E85", LBA Type Certificate No. 60.230/1
Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P
"Europa G 88" safety tow release LBA Type Certificate No. 60.230/2
6. Weak links: Ultimate Strength:
- for winch launching max. 6230 N
- for aero-tow max. 6230 N
7. Air Speeds: Air speeds of sailplane without wing tip extensions:
Manoeuvring Speed V_A 160 km/h IAS
Never Exceed Speed V_{NE}
up to 2500 m MSL 230 km/h IAS
2500 to 3000 m MSL 223 km/h IAS
3000 to 4000 m MSL 209 km/h IAS
4000 to 5000 m MSL 195 km/h IAS
5000 to 6000 m MSL 182 km/h IAS
6000 to 7000 m MSL 170 km/h IAS
7000 to 8000 m MSL 158 km/h IAS
8000 to 9000 m MSL 147 km/h IAS

| | |
|--|--------------|
| 9000 to 10000 m MSL | 136 km/h IAS |
| Rough Air Speed V_{RA} | 160 km/h |
| Max. Aerotow Speed V_T | 150 km/h |
| Max. Winch-launch Speed V_W | 120 km/h |
| Max. Landing Gear Operating Speed V_{LO} | --- |

| | |
|--|---|
| | Air speeds of sailplane with wing tip extensions: |
| Maneuvering Speed V_A | 150 km/h IAS |
| Never Exceed Speed V_{NE} | |
| up to 2500 m MSL | 230 km/h IAS |
| 2500 to 3000 m MSL | 223 km/h IAS |
| 3000 to 4000 m MSL | 209 km/h IAS |
| 4000 to 5000 m MSL | 195 km/h IAS |
| 5000 to 6000 m MSL | 182 km/h IAS |
| 6000 to 7000 m MSL | 170 km/h IAS |
| 7000 to 8000 m MSL | 158 km/h IAS |
| 8000 to 9000 m MSL | 147 km/h IAS |
| 9000 to 10000 m MSL | 136 km/h IAS |
| Rough Air Speed V_{RA} | 150 km/h |
| Max. Aerotow Speed V_T | 150 km/h |
| Max. Winch-launch Speed V_W | 120 km/h |
| Max. Landing Gear Operating Speed V_{LO} | --- |

- | | | | |
|--|--|-------------|--------------|
| 8. Operational Capability: | VFR Day Cloud flying | | |
| 9. Maximum Weights: | | | |
| Maximum Weight: | 500 kg (without extensions) 510 kg (with extensions) | | |
| Maximum Weight of non-lifting parts: | 355 kg | | |
| Empty Weight: | 305 kg \pm 2% (without extensions) 315 kg \pm 2% (with extensions) | | |
| 10. Centre of Gravity Range: | Fore most c.g. limit aft of reference plane 143 mm Aft most c.g. limit aft of reference plane 337 mm [MAC is 1295 mm without wing tip extensions] [MAC is 1252.3 mm with wing tip extensions] | | |
| 11. Datum: | Wing leading edge at root rib | | |
| 12. Levelling Means: | Leveling points on fuselage in horizontal position. | | |
| 13. Minimum Flight Crew: | 1 (Pilot) | | |
| 14. Maximum Passenger Seating Capacity: | 1 | | |
| 15. Lifetime limitations: | Refer to Maintenance Manual | | |
| 16. Deflection angles of control surfaces: | | | |
| | Elevator | up | 32° + 2° |
| | | down | 27° \pm 1° |
| | Rudder | left, right | 29° + 1° |
| | Ailerons | up | 34° + 2° |
| | | down | 13° + 2° |
| | Left aileron balance tab | up | 20° \pm 2° |
| | | down | 15° \pm 2° |
| | Elevator trim tab | up | 12° \pm 1° |
| | | down | 35° \pm 1° |

B.IV. Operating and Service Instructions

1. Flight Manual:

| | |
|--|--|
| In Czech language Do-L13AC-1013.1 | Letová příručka L 13 AC Blaník (to S/N 008606 from S/N 028902 to 029101) |
| Do-L13AC-1014.0 | Letová příručka L 13 AC Blaník (S/N 018901 from S/N 039102) |
| In English language Do-L13AC-1013.3 | Sailplane Flight Manual L 13AC Blaník to S/N 008606 from S/N 028902 to 029101 |
| Do-L13AC-1014.2 | Sailplane Flight Manual L 13AC Blaník (S/N 018901 from S/N039102) |

2. Maintenance Manual:

| | |
|--|--|
| In Czech language Do-L13AC-1032.1 | Provozně technická příručka kluzáku L 13 AC |
| In English language Do-L13AC-1032.3 | Maintenance Manual for the Sailplane L 13 AC |

3. Illustrated Parts Catalogue:

| | |
|--|---|
| In Czech language Do-L13AC-2051.0 | Katalog dílů a montážních jednotek kluzáku L-13AC Blaník (C/A) |
| In English language Do-L13AC-2051.0 | Illustrated parts catalogue for the sailplane L 13 AC Blaník (C/A) |

4. Operation book

| | |
|---|--|
| In Czech language Do-L13AC.1015.02 | L 13AC Blaník Záznamník provozu kluzáku (C/A) |
| In English language Do-L13AC.1015.02 | L 13 AC Blaník Sailplane operation book of records (C/A) |

5. Operating Manuals for Tow Releases

| | |
|-----------------------------------|---|
| In German and in English language | Operating Manual for Nose Tow Releases TOST "Europa E 85" |
| | Operating Manual for Safety Tow Releases TOST "Europa G 88" |

B.V. Notes

1. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.
2. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.
3. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.

C.I. General (L 13 A Blaník)

| | | |
|--|--|-------------------------|
| Data Sheet No.: EASA.A.024 | Issue: 10 | Date: 13 September 2019 |
| 1. Model: | L 13 A Blaník | |
| 2. Airworthiness Category: | Acrobatic Cloud flying | |
| 3. Type Certificate Holder: | Blanik Aircraft CZ s.r.o. Beranových 65 190 00 Praha 9 - Letňany Czech Republic | |
| 4. Manufacturer: | from S/N 817401 to S/N 827420 LET, n.p. 686 04 Kunovice 1177 CZECH REPUBLIC from S/N 968501 to S/N 968505 and S/N 827421 LET, a.s. 686 04 Kunovice 1177 CZECH REPUBLIC | |
| 5. Certification Application Date: | --- | |
| 6. CAA CZ Type Certification Date: | December 16, 1981 | |
| 7. EASA Type Certification Date: | 12 August 2005 | |
| 8. The EASA Type Certificate replaces Czech Type Certificate No. 2725-59 | | |

C.II. Certification Basis

| | |
|--|--------------------------------------|
| 1. Reference Date for determining the applicable requirements: | --- |
| 2. Certification Basis: | BCAR, Section E, issued June 6, 1966 |
| 3. Airworthiness Requirements: | see (2) above |
| 4. Requirements elected to comply: | None |
| 5. EASA Special Conditions: | None |
| 6. EASA Exemptions: | None |
| 7. EASA Equivalent Safety Findings: | None |

C.III. Technical Characteristics and Operational Limitations

- | | |
|--|--|
| 1. Type Design Definition: | Drawing No. A 101 310 N |
| 2. Description: | All-metal, cantilever, high-wing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake, and a tail skid or (an optional) tail wheel. Horizontal tail surfaces consist of a two-piece tail-plane and elevator, vertical tail surfaces consist of a fin and a rudder. |
| 3. Equipment: | Minimum equipment: 2 airspeed indicators, range to 400 km/h 2 altimeters 2 four-point safety harnesses (symmetric) 2 parachutes or backrests (approx. 10 cm thick when compressed) |
| 4. Dimensions: | |
| Span | 16.2 m |
| Length | 8.4 m |
| Height | 2.1 m |
| Wing Area | 19.15 m ² |
| Aspect Ratio | 13.7 |
| 5. Launching Hooks: | Nose tow release of type SR-L13.225, or nose tow release Dwg. No. A 740 210 N, or nose tow release "E85", LBA Type Certificate No. 60.230/1 Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P "Europa G 72" safety tow release LBA Type Certificate No. 60.230/2, or "Europa G 73" safety tow release LBA Type Certificate No. 60.230/2, or "Europa G 88" safety tow release LBA Type Certificate No. 60.230/2 |
| 6. Weak links: | Ultimate Strength: - for winch launching max. 6230 N - for aero-tow max. 6230 N |
| 7. Air Speeds: | |
| Manoeuvring Speed V_A | 145 km/h IAS |
| Never Exceed Speed V_{NE} | |
| up to 2500 m MSL | 253 km/h IAS |
| 2500 to 3000 m MSL | 245 km/h IAS |
| 3000 to 4000 m MSL | 230 km/h IAS |
| 4000 to 5000 m MSL | 215 km/h IAS |
| 5000 to 6000 m MSL | 201 km/h IAS |
| 6000 to 7000 m MSL | 187 km/h IAS |
| 7000 to 8000 m MSL | 174 km/h IAS |
| 8000 to 9000 m MSL | 161 km/h IAS |
| 9000 to 10000 m MSL | 150 km/h IAS |
| Rough Air Speed V_{RA} | 145 km/h |
| Max. Aerotow Speed V_T | 140 km/h |
| Max. Winch-launch Speed V_W | 120 km/h |
| Max. Landing Gear Operating Speed V_{LO} | --- |

| | | | |
|--|---|-------------|----------|
| 8. Operational Capability: | VFR Day Cloud flying | | |
| 9. Maximum Weights: | | | |
| Maximum Weight: | 500 kg | | |
| Maximum Weight of non-lifting parts: | 340 kg | | |
| Empty Weight: | 306 kg ± 2% | | |
| 10. Centre of Gravity Range: | Fore most c.g. limit aft of reference plane 112 mm Aft most c.g. limit aft of reference plane 300 mm [MAC is 1253 mm] | | |
| 11. Datum: | Wing leading edge at root rib | | |
| 12. Levelling Means: | Leveling points on fuselage in horizontal position. | | |
| 13. Minimum Flight Crew: | 1 (Pilot) | | |
| 14. Maximum Passenger Seating Capacity: | 1 | | |
| 15. Lifetime limitations: | Refer to Maintenance Manual | | |
| 16. Deflection angles of control surfaces: | | | |
| | Elevator | up | 32° + 2° |
| | | down | 25° ± 1° |
| | Rudder | left, right | 30° + 1° |
| | Ailerons | up | 34° + 2° |
| | | down | 13° + 2° |
| | Wing flaps | down | 8° ± 1° |
| | Elevator trim tab | up | 12° ± 1° |
| | | down | 35° ± 1° |

C.IV. Operating and Service Instructions

1. Flight Manual:

| | |
|---------------------------------------|---------------------------------------|
| In Czech language Do-L13A-1011.1 | Letová příručka L 13 A |
| In English language Do-L13A-1011.3 | Flight Manual of the L 13 A |
| In German language Do-L13A-1011.2 | Flughandbuch für das Segelflug L 13 A |

2. Maintenance Manual:

| | |
|---------------------------------------|--|
| In Czech language Do-L13A-1031.1 | Příručka pro obsluhu, údržbu a opravy kluzáku L 13 |
| In English language Do-L13A-1031.3 | Technical Manual of the L 13 A Sailplane |
| In German language Do-L13A-1031.2 | Handbuch für die Reparatur, Instandhaltung und Wartung des Segelflugzeuges L 13 A |

3. Illustrated Parts Catalogue:

| | |
|--------------------------------------|--|
| In Czech language Do-L13-2126.6 | Kusovník větroně L 13 Blaník (C-A-N) |
| In English language Do-L13-2126.6 | Spare Parts Catalogue L 13 Blaník (C-A-N) |
| In German language Do-L13-2126.6 | Katalog der Bestandteile L-13 Blaník (C-A-N) |

4. Overhaul Manual

| | |
|--------------------------------------|---|
| In Czech language Do-L13-3031.1 | Příručka pro generální opravu kluzáku L 13, L 13A |
| In English language Do-L13-3031.3 | Overhaul Manual for L 13, L 13A Gliders |

5. Sailplane Technical Description L 13A

| | |
|-------------------------------------|---------------------------------------|
| In Czech language Do-L13A-1021.1 | Sailplane Technical Description L 13A |
|-------------------------------------|---------------------------------------|

6. Operating Manuals for Tow Releases

| | |
|-----------------------------------|---|
| In German and in English language | Operating Manual for Nose Tow Releases TOST "Europa E 85" |
| | Operating Manual for Safety Tow Releases TOST "Europa G 88" |
| | Operating Manual for Safety Tow Releases TOST "Europa G 72" and "Europa G 73" |

C.V. Notes

1. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.
2. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.
3. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.

ADMINISTRATIVE SECTION

I. Acronyms

N/A

II. Type Certificate Holder Record

Up to 19 June 2013

Aircraft Industries, a.s.
Na Záhonech 1177
686 04 Kunovice
Czech Republic

Up to 09 January 2017

BLANIK LIMITED
2nd Floor Beaux Lane House
Mercer Street Lower
Dublin 2
Republic of Ireland

Up to 12 September 2019

Blanik Aircraft CZ s.r.o.
Karolinská 661/4
186 00 Praha 8
Czech Republic

Since 13 September 2019

Blanik Aircraft CZ s.r.o.
Beranových 65
190 00 Praha 9 - Letňany
Czech Republic

III. Change Record

| Issue | Date | Changes |
|--------------|-------------------|--|
| 01 | 04 February 2005 | Initial issue of TCDS No. EASA.A.044 |
| 02 | 12 August 2005 | To record the change in the name of the TC holder and to include the L 13 A model. |
| 03 | 09 September 2005 | Editorial changes |
| 04 | 23 May 2006 | Change in address of the TC holder |
| 05 | 20 June 2013 | Change of the TC holder and new layout of TCDS |
| 06 | 09 May 2016 | Addition of S/N 827421 (L 13 A Blanik) to section C.I.5. |
| 07 | 21 September 2016 | Addition of CS 22.627 Eltect to Comply to section A.II.4. and addition of Sailplane Flight Manual (Do-L13-1111.6) to section A.IV.1. |
| 08 | 30 September 2016 | Change of the Contracted DOA |
| 09 | 10 January 2017 | Change of the TC holder |
| 10 | 13 September 2019 | Change of the TC holder's address |