WT9 Dynamic LSA Date: 30 October 2020 TCDS N.: EASA.A.644

Issue: 04



TYPE-CERTIFICATE DATA SHEET

NO. EASA.A.644

for WT9 Dynamic LSA

Type Certificate Holder Aerospool, spol. s r. o.

Letisková 10, 971 03 Prievidza Slovak Republic

For models: Club

Issue: 04

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MODEL A DESIGNATION4

SECTION A:

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SECTION A: MODEL A DESIGNATION

A.I. <u>General</u>

1. Type/ Model/ Variant

1.1 Type: WT9 Dynamic LSA

1.2 Model: Club

2. Airworthiness Category: Normal (see Note 2)

3. Manufacturer: Aerospool, spol. s r. o.

Letisková 10, 971 03 Prievidza

Slovak Republic

4. EASA Type Certification Application Date:21.12.20105. EASA Type Certification Date:31.08.2017

A.II. EASA Certification Basis

1. Reference Date for determining the applicable requirements: 31.12.2014

2. Airworthiness Requirements: Certification Specifications and

Acceptable Means of Compliance for Light Sport Aeroplanes CS-LSA, Amendment 1 from 29 July 2013. For detailed information see CRI-

A01

3. Special Conditions: none

4. Exemptions: none

5. (Reserved) Deviations: none

6. Equivalent Safety Findings: ELOS-BLSA.0221-01 - Spinning (CRI B-54),

7. Environmental Protection: CS-36, Amendment 3 (CRI N-01)

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A.III. Technical Characteristics and Operational Limitations

1. Type Design Definition: Type Design Definition, AS-RTC-15-001.A1 revision 12 or later

approved issue

2. Description: Single engine, two-seat, low wing composite construction aircraft

with conventional empennage configuration and fixed tricycle

landing gear.

3. Equipment: Minimum equipment, see Pilot's Operating Handbook,

AS-POH-01-000, Initial issue 15.05.2017 or later approved issue,

Section 2.13.3

4. Dimensions

5. Engine

5.1. Model: Rotax 912 ULS2

5.2 Type Certificate: Certified as part of the aircraft

5.3 Limitations: Maximum Power Rating: 73.5 kW / 5800 RPM max. 5 min

Maximum Continuous Power: 69 kW / 5500 RPM

6. Load factors: +4G/-2G

7. Propeller Option 1:

Model: EVRA Performance Line 175/xxx/805.5

Type Certificate: Certified as part of the aircraft

Number of blades: 3

Diameter: 1750 ±3 mm

Rotation direction: Clockwise (in flight direction)

Option 2 (configuration FG912T, see Note 1):

Model: Aleš Křemen / KW-31

Type Certificate: EASA.P.177

Number of blades: 3

Diameter: 1726 ±4 mm

Rotation direction: Clockwise (in flight direction)

8. Fluids

8.1 Fue

Refer to Pilot Operating Handbook AS-POH-01-000, Initial issue 15.05.2017 or later approved issue, Section 2

8.2 Oil



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Refer to Pilot Operating Handbook AS-POH-01-000, Initial issue 15.05.2017 or later approved issue, Section 2

8.3 Coolant

Refer to Pilot Operating Handbook AS-POH-01-000, Initial issue 15.05.2017 or later approved issue, Section 2

9. Fluid capacities

9.1 Fuel Total: 126 liters

Usable: 119 liters

9.2 Oil Maximum oil capacity: 3.5 liters

Minimum oil required: marked on dipstick

9.3 Coolant system 2.5 liters (approximately)

10. Air Speeds (CAS) V_{NE}: 142 kts (263 km/h)

V_c: 113 kts (210 km/h)

V_c: 119 kts (220 km/h) (configuration FG912T, see Note 1)

V_A: 96 kts (177 km/h) V_F: 83 kts (153 km/h)

11. Flight Envelope Maximum operating altitude: 15,000 ft MSL

(17,800 ft MSL for configuration FG912T, see Note 1)

12. Approved Operations

Capability

VFR day operations only

13. Maximum Masses Maximum takeoff: 600 kg

Maximum landing: 600 kg Maximum empty: 410,8 kg Maximum fuel: 90,7 kg

14. Centre of Gravity Range Forward CG limit: 18,3% MAC / 2704 mm aft of datum

Aft CG limit: 28.5% MAC / 2824 mm aft of datum

15. Reference datum 1975 mm forward of firewall (surface from cockpit)

16. Control surface deflections Refer to AMM

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17. Levelling Means AS-POH-01-000, Initial issue 15.05.2017 or later approved issue,

Section 6

AS-AMM-01-000, Initial issue 22. 05. 2017 or later approved

issue, ATA Chapter 08

18. Minimum Flight Crew One (1) pilot

19. Maximum Passenger

Seating Capacity

One (1) passenger

20. Baggage/ Cargo Forward compartment: divided in 2 sections
Compartments Maximum load: 20 kg (10 kg in each section)

Rear compartment: behind backrest, divided in 2 sections

Maximum load: 40 kg (20 kg in each section)

21. Wheels and Tyres Main wheel Tyre Size 15x6.00-6

Nose wheel Tyre Size 13x5.00-6

22. Lifetime limitations Refer to AMM, AS-AMM-01-000, Initial issue 22.05.2017 or later

approved issue, Section 05

A.IV. Operating and Service Instructions

1. Flight Manual AS-POH-01-000, Initial issue 15.05.2017 or later approved

issue

2. Maintenance Manual AS-AMM-01-000, Initial issue 22.05.2017 or later approved

issue

3. Structural Repair Manual Refer to AMM

4. Weight and Balance Manual Refer to POH

5. Wiring Manual AS-AWM-01-000, Initial issue 07.08.2017 or later approved

issue

6. Illustrated Parts Catalogue AS-IPC-01-000, Initial issue 07.08.2017 or later approved

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A.V. Notes

1) Configuration FG912T is equipped with Aleš Křemen KW-31 propeller and the tow equipment, consisting of tow release, rear camera and rear mirror. EASA Noise Record C14067 is applicable.

2) Serial numbers produced before 30 October 2020 are eligible for Restricted Certificate of Airworthiness in accordance with TCDS EASA.A.644 issue 03 and previous. Serial number produced after 30 October 2020 are eligible for normal Certificate of Airworthiness in accordance with this TCDS EASA.A.644 issue 04.

SECTION ADMINISTRATIVE

I. Acronyms & Abbreviations

AMM Aircraft maintenance manual CS-LSA Certification specification for light sport aeroplanes EASA European aviation safety agency IPC Illustrated parts catalogue KIAS Indicated airspeed in knots **KTAS** True airspeed in knots MAC Mean aerodynamic chord MSI Mean sea level MDL Master document list POH Pilot's operating handbook **RPM** Revolutions per minute

VFR Visual flight rules

II. Type Certificate Holder Record

Aerospool, spol. s r. o. Letisková 10, 971 03 Prievidza Slovak Republic

III. Change Record

Issue	Date	Changes	TC Issue No. & Date
Issue 01	31/08/2017	Initial Issue	31/08/2017
Issue 02	26/07/2019	New propeller KW-31 added. Change of maximum operating altitude for towing configuration FG912T. Note 1 added. Administrative changes.	31/08/2017
Issue 03	30/04/2020	Administrative correction of a designator of Designed Flap Speed $V_{\scriptscriptstyle F}$	31/08/2017
Issue 04	30/10/2020	Change of airworthiness category from Restricted to Normal in Section A.I.2. Addition of Note A.V.2	30/10/2020