



EASA
European Aviation Safety Agency

Elastomeric parts

Herdrice HERESON

EASA Structures Expert

Rotorcraft Structures Workshop
19-20 February 2019





Elastomers are widely used in Industry



Bridges



Airplanes



Helicopters

1920s

- Hugh Lord forms LORD Manufacturing Company.

1930s

- Jean-Félix Paulsen forms Paulstra to designing & developing anti-vibration mountings & systems on Aircraft

1940s

- LORD first elastomeric parts – transmission and engine isolators for the Bell Model 47 helicopter

1950s

- LORD create SPE and BTR elastomers

1970s

- Hutchinson buy Paulstra/ Total is Hutchison major staholder
- LORD Elastomeric bearings acceptance in the helicopter industry

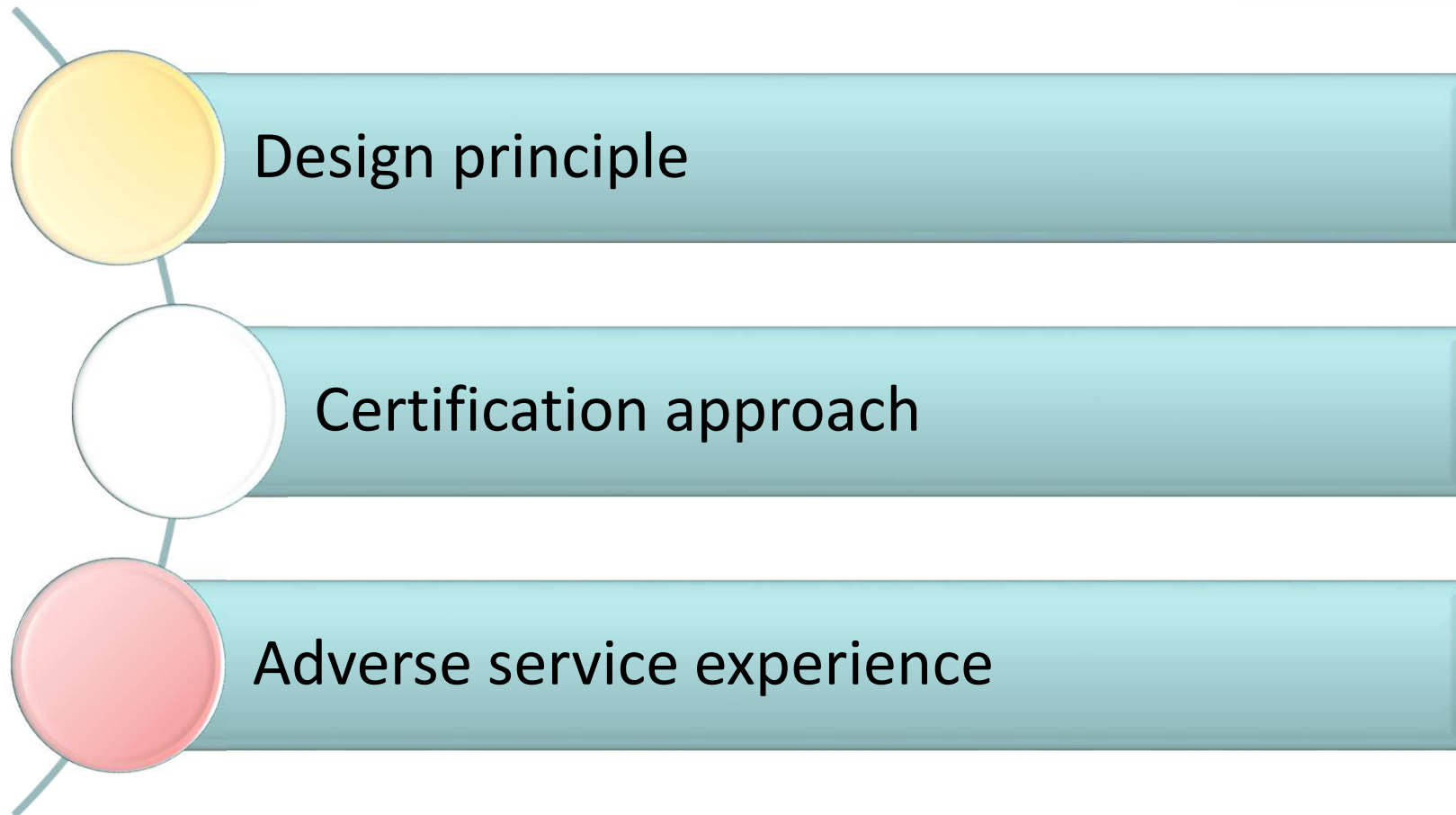
1990s

- Paulstra's acquires Vibrachoc(specialize d business on metallic suspension)
- LORD equip Russian helicopters and fixed wing aircraft
-



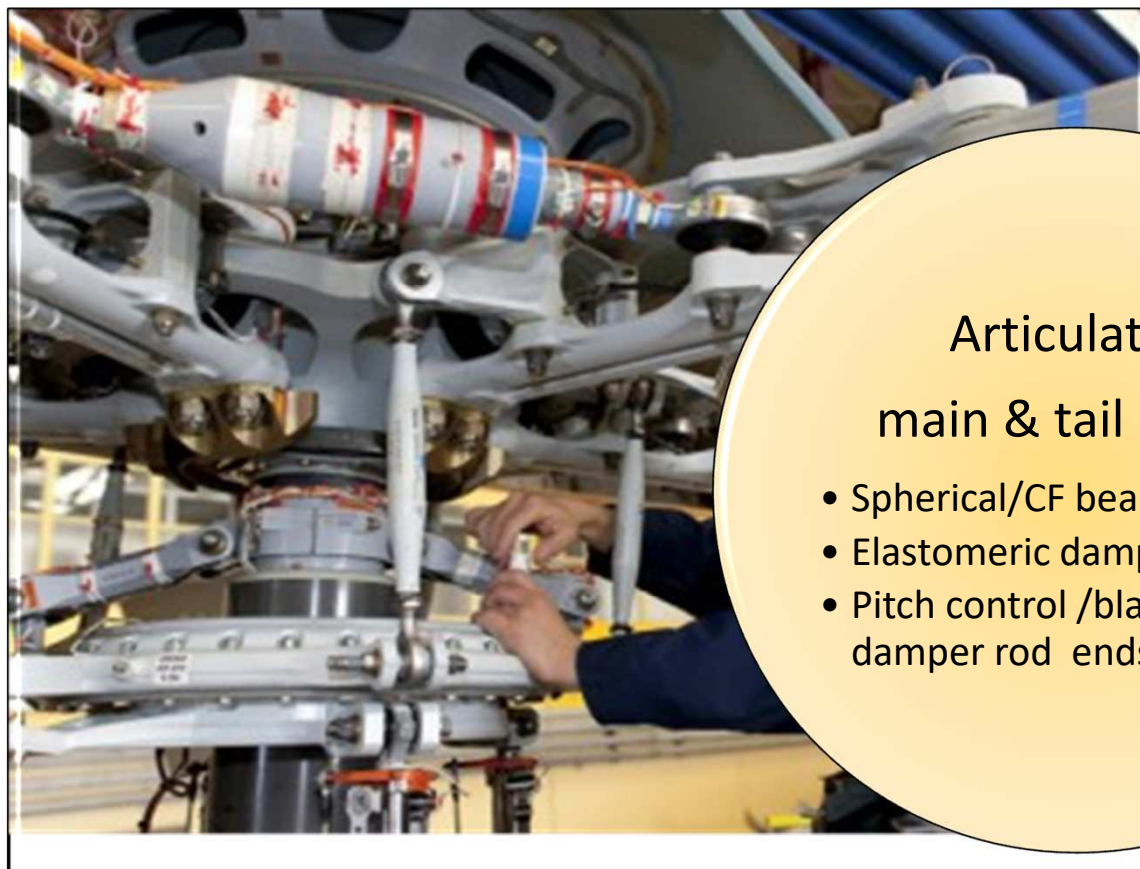


Presentation objectives:





Elastomers on helicopters increased since 2010



Articulated main & tail rotors

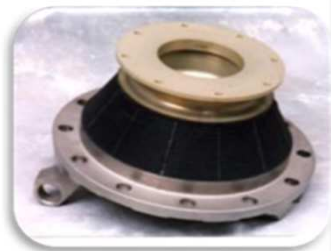
- Spherical/CF bearing
- Elastomeric damper
- Pitch control /blade damper rod ends

Airframe

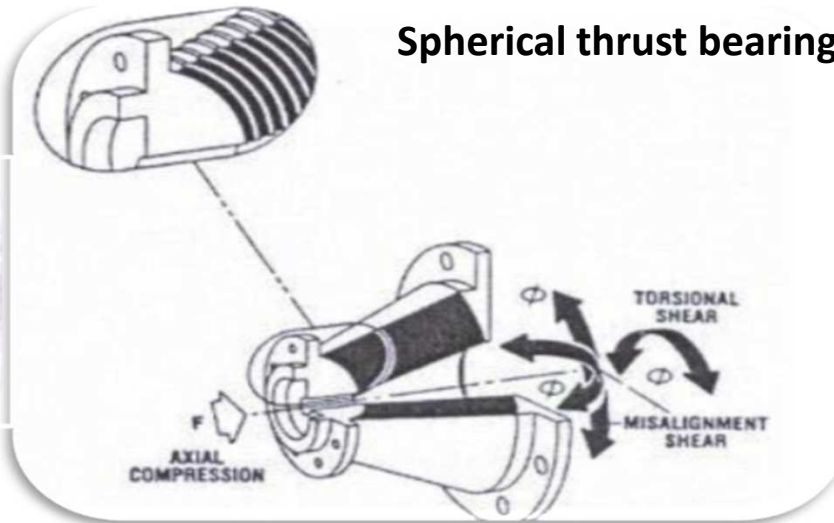
- MGB suspension bearings
- Engine mount
- AVCS



Type of Elastomer used on Rotorcraft: Laminates



Spherical thrust bearing



Elastomeric compound
(LORD SPE, Paulstra confidential)

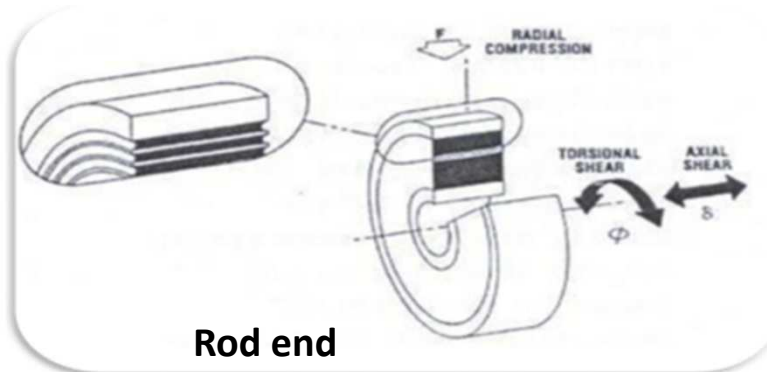
- High Elastic Solid
- Incompressible (Poisson ratio is about 0,5)

Metal Shim (Al, steel, Ti..)

- Act as « races » of conventional bearings
- Mechanism to apply and react loads and motion



Rod end



- No abrasive wear.
- No lubrication required
- Fail-safe design/capability
- Simple maintenance.
- Capacity of tuning system to natural frequency



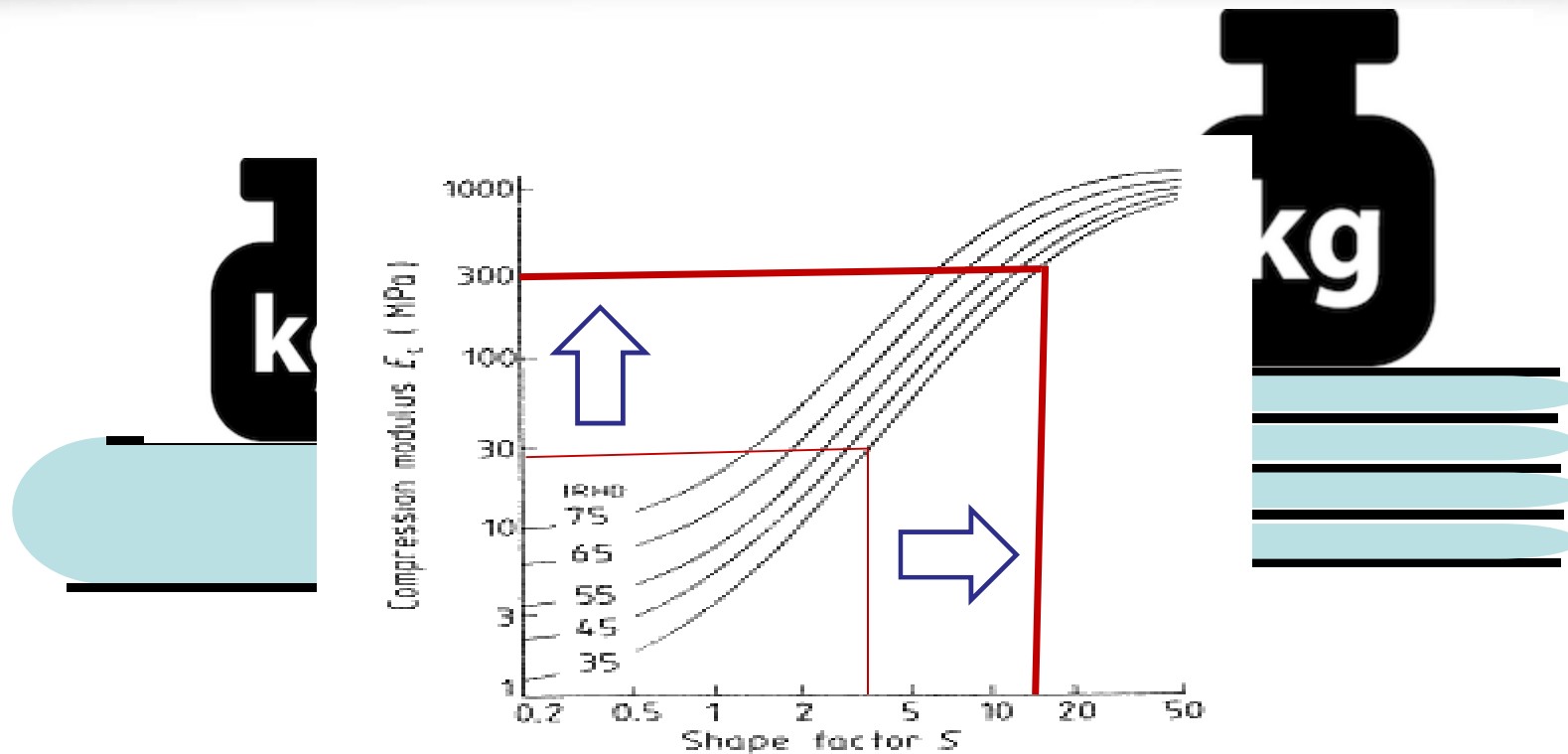
Illustration of laminate performance

Same load surface





Illustration of laminate performance

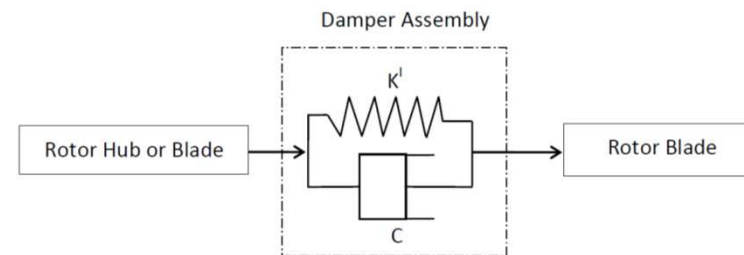


• $S = f(\text{compression surface} / \text{rubber thickness per layer})$.



Type of Elastomer used on Rotorcraft: *Rubber (silicone)*

Elastomeric damper



•Example: LORD BPR II

•**BPR®II** :Broad Temperature Range- Silicone elastomer developed to have high Damping and a wide span of operational temperatures (-40°C to +149°C). The loss factor for typical BTR II compounds is in the range of 0.18. This elastomer has better returnability, less drift and better stability with temperature, down to -40°F (-40°C)..



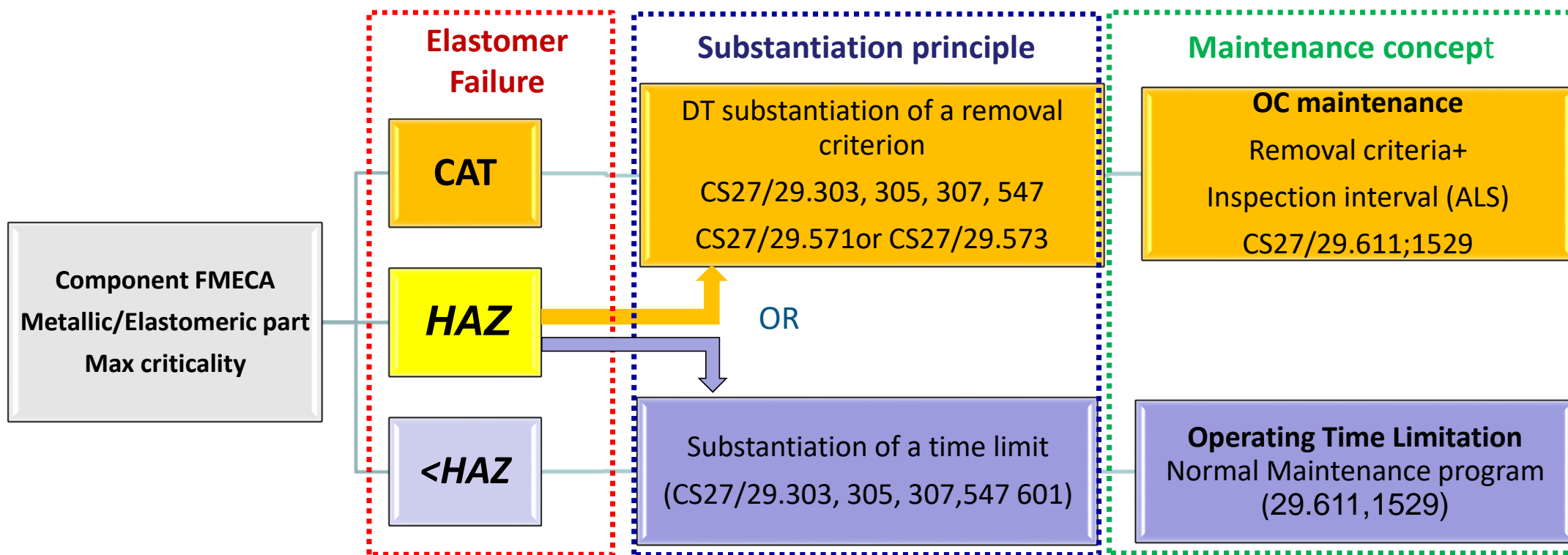
Presentation objectives:





Elastomeric parts have different criticalities

CS27/29 Rotorcraft To avoid catastrophic failure

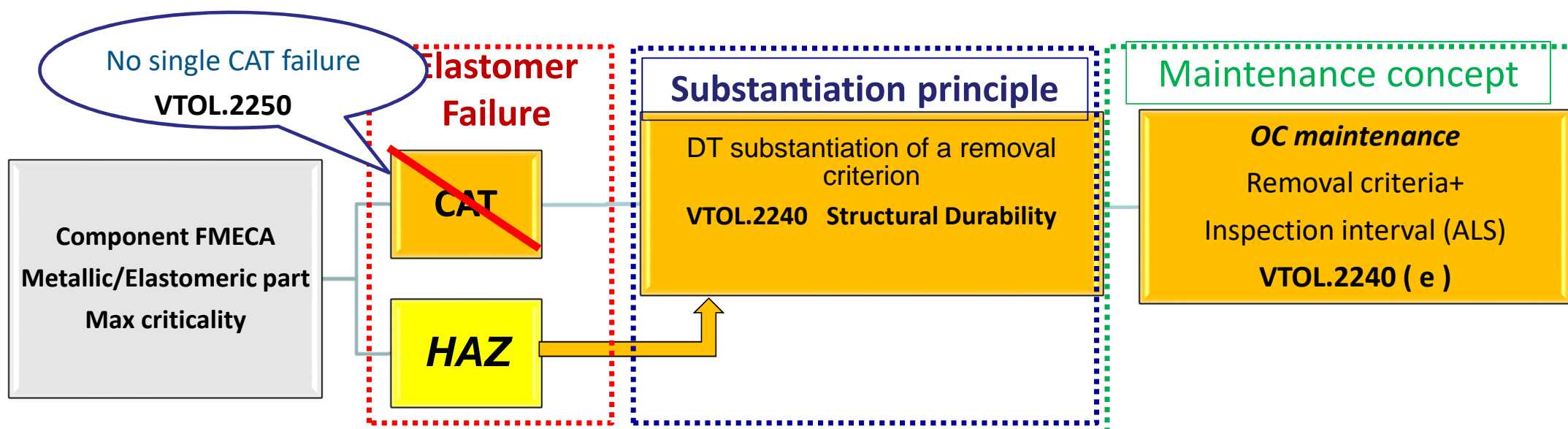




Elastomeric parts have different criticalities

SC-VTOL Enhanced

VTOL.2240-The applicant must develop and implement inspections or other procedures to prevent structural failures due to foreseeable causes of strength degradation, which could result in serious or fatal injuries, or extended periods of operation with reduced safety margins





Supplier/OEM-Roles and responsibilities

•Elastomer Supplier tests

Qualification test (DT test)

Elastomer Manufacturing process
&Critical characteristics control
(CS27/29.602, 609..)

Characterization tests

- Temperature
- Heat aged (ASTM573)
- Tensile strength and Elongation (ASTMD412)
- Compression (ASTMD395)
- Static modulus Dynamic modulus
- Ozone (ASTM1149)
- Fluids (oils, fuels, coolants, etc.) & the level of fluid exposure (immersion vs. splash)
- (...)

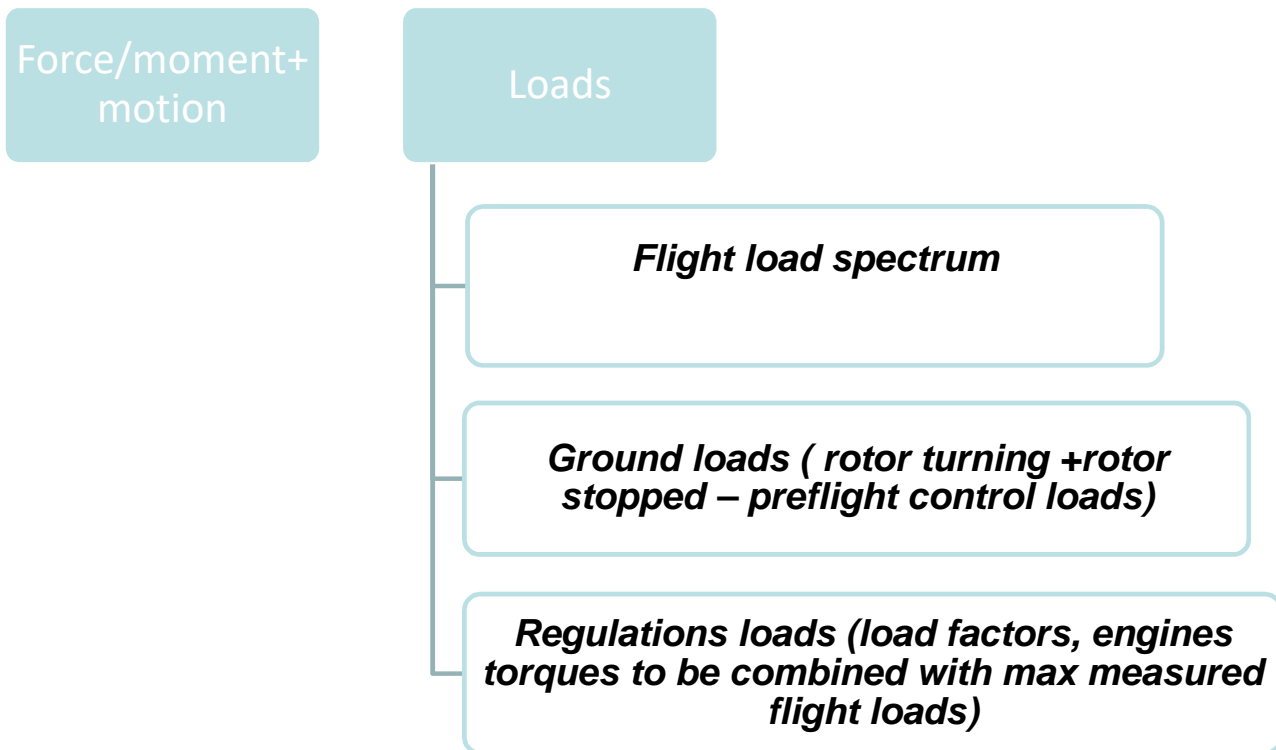
Industrial property
make OEM control
difficult

•OEM Tests/Evaluation

- Instrumented flight tests
- Continued safe flight and landing test/ analysis after bird impact/lightning strike)
- Vibration/Ground resonance/Aeroelastic stability evaluation (nominal and with reduced stiffness)
- Any other applicable requirements

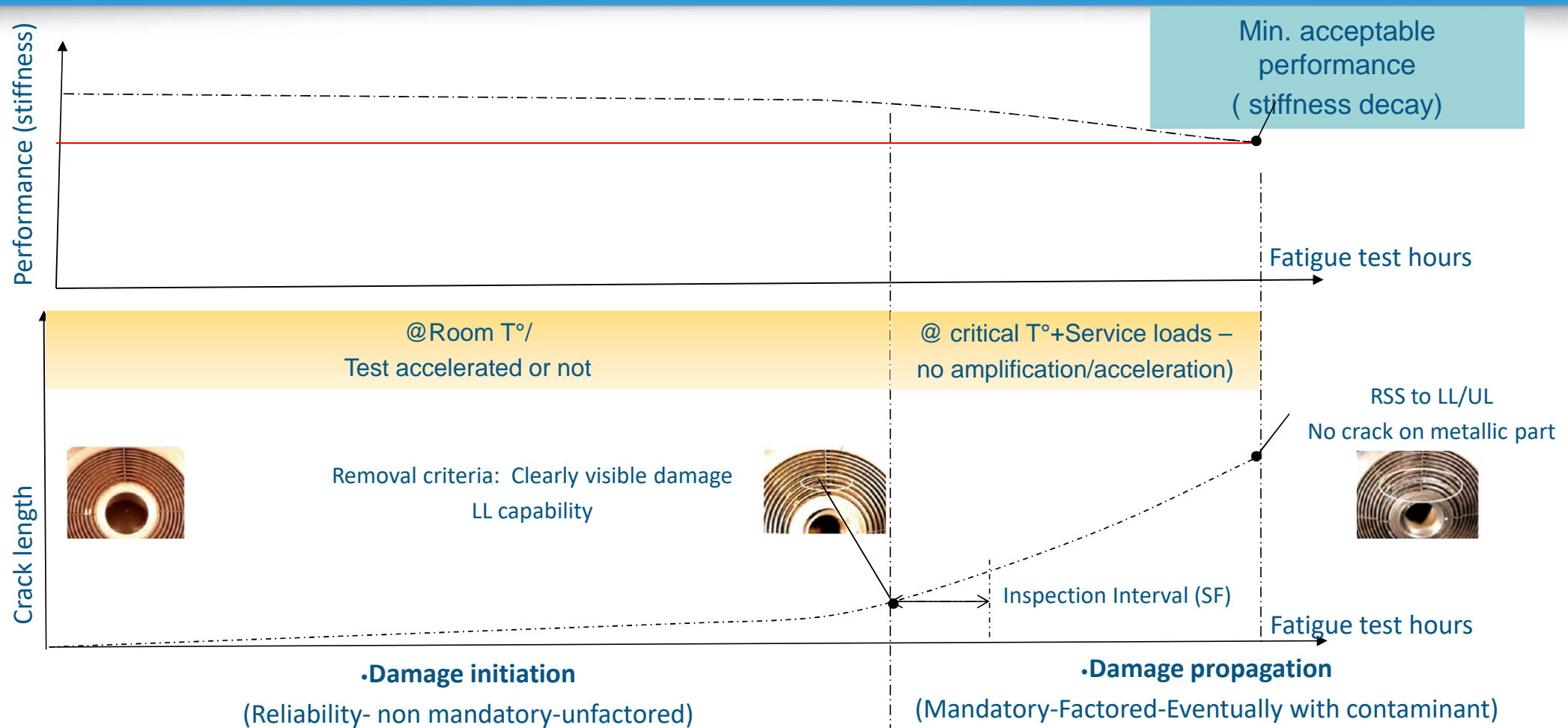


Major input-Qualification Test Spectrum



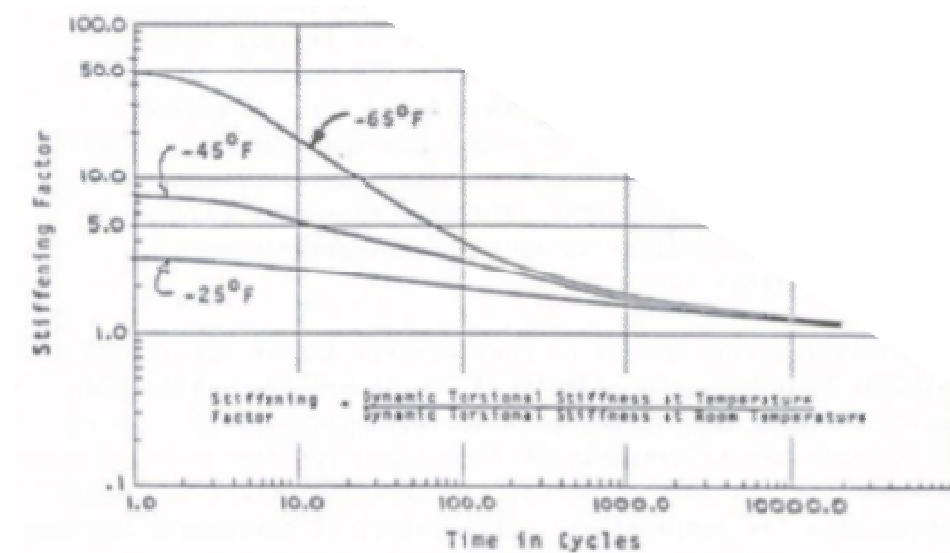
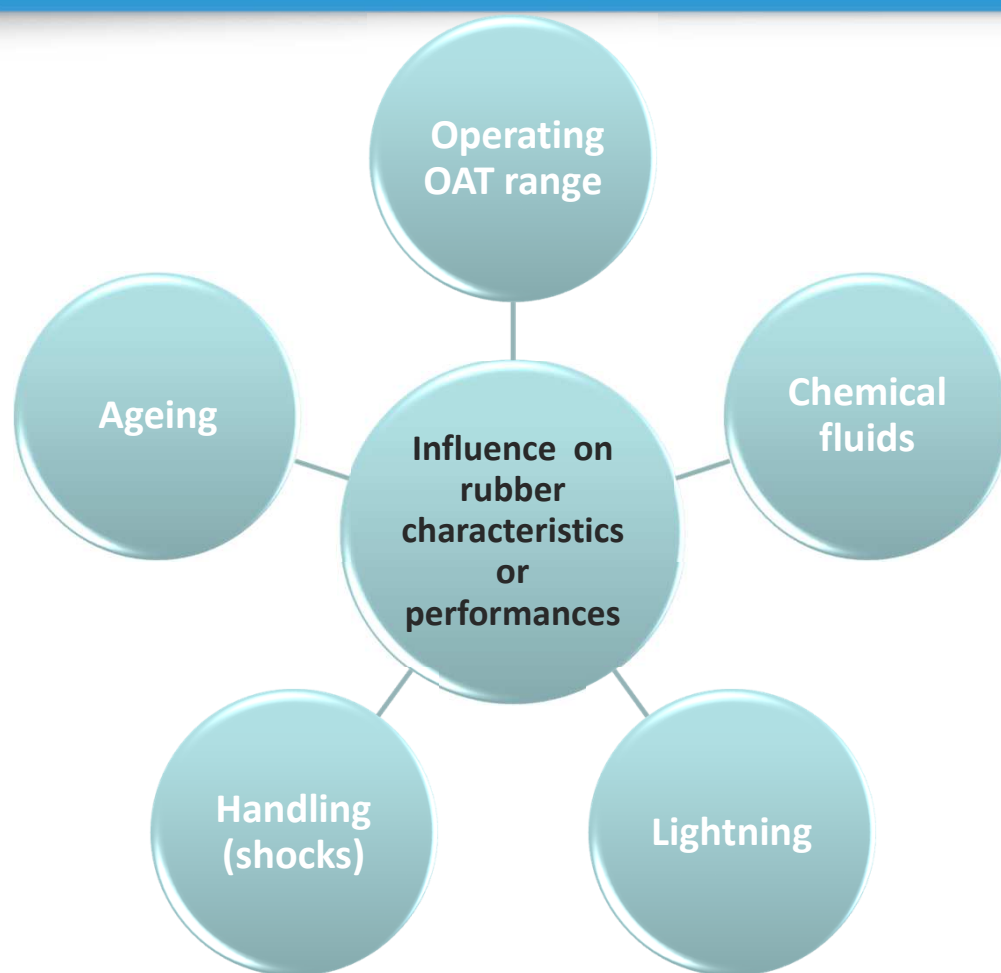


Qualification test –Damage tolerance fatigue test





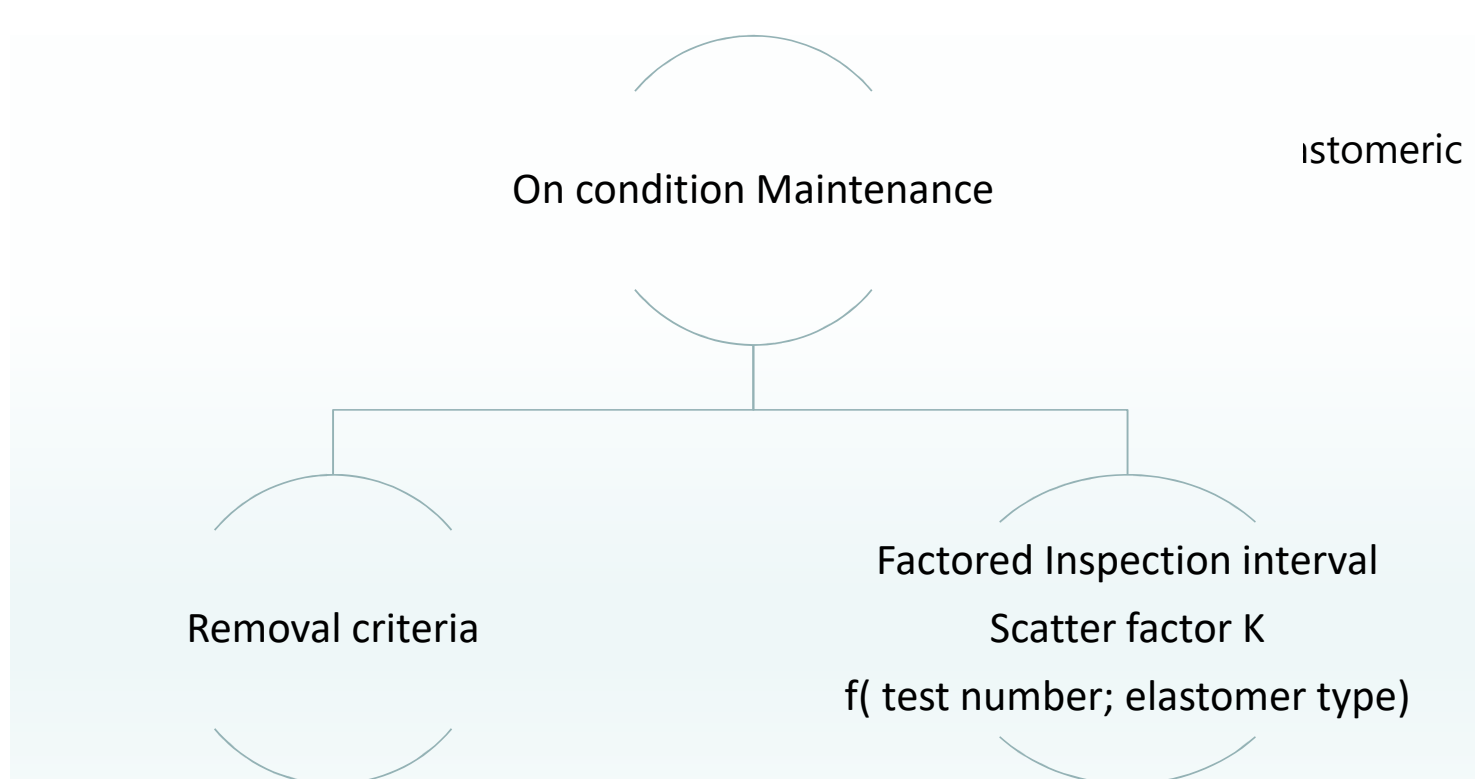
Main parameters influencing performance





Inspection criteria/Degradation mode

• <https://www.youtube.com/watch?v=3tSHzSLh3Yw>



Elastomeric Helicopter Parts (4



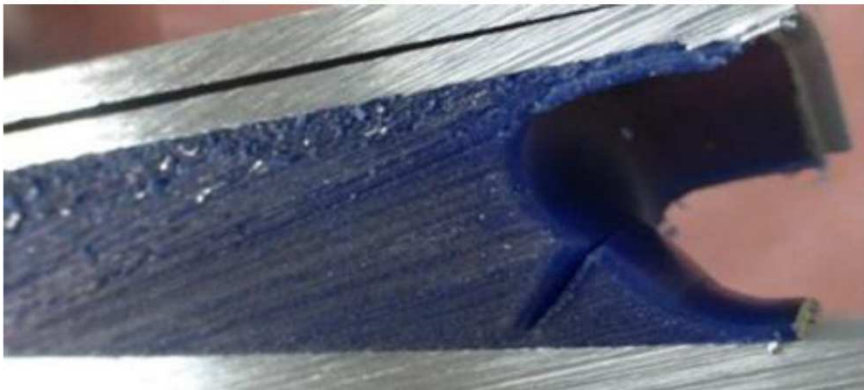
Presentation objectives:





Adverse service experience 1

159 Main rotor damper discarded - Removal criteria reached after 12FH min. instead of 55FH inspection (2000 FH initial reliability)



Root cause

The adjustment of a parameter of the damper production process have “polluted the raw material”. Not detected with the supplier control checks



Adverse service experience2

Tail rotor Laminated bearing failure- Fatal accident (TSN 101h-post MOD configuration)



Root cause

Non-conservative spectrum
Addition of Chinese weight (passive control
load compensator)/Design issue



Adverse service experience 3

Main rotor damper link elastomeric bearing failure (TSN: 75FH)

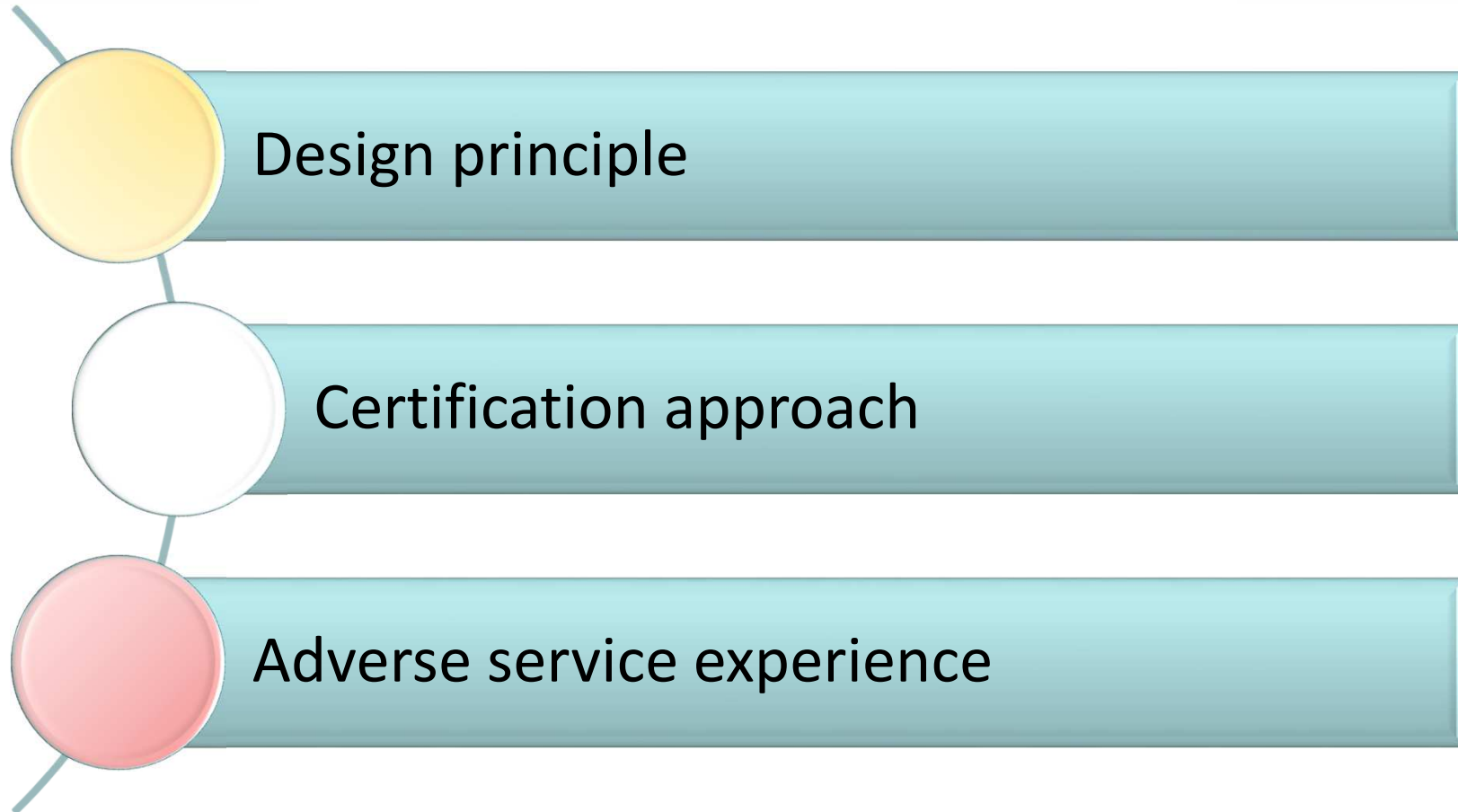


Root cause

No test performed at low temperature
(penalty factor instead)



Presentation objectives:





Conclusion on elastomeric components

Improved compared to normal bearings & easy to maintain.

More and more extended!

But still more challenges

- No dedicated elastomeric parts requirement
- Classification difficult (impact @ rotorcraft level)
- OEM highly dependant on the supplier (CS27/29-602-Critical process change & control)
- Spectrum & environmental conditions sensitive
- Scatter factor are not well harmonized
- Detectability (Room for improvement)



EASA
European Aviation Safety Agency

Thank you for your attention!

Any questions....?

Your safety is our mission.

An agency of the European Union

