**What are the component release documents acceptable for installing such component (on aircraft or on a higher assembly) and releasing associated maintenance under EASA Part-145 regulation?**

This table is provided FOR REFERENCE ONLY. The acceptance of these release documents may be subject to specific conditions expressed in the applicable regulation or agreement (e.g. TIP/MAG) on EASA website.

**Definitions:**
- For the purpose of this FAQ, the following definitions are used:
  - 'Component' means any engine, propeller, part or appliance (Article 2(c) of Reg. (EU) 1321/2014)
  - SINGLE release means the component maintenance has been released under one regulatory framework only
  - DUAL release means the component maintenance has been released concurrently under two regulatory frameworks
  - TRIPLE release means the component maintenance has been released concurrently under three regulatory frameworks (today limited to EASA Form 1 released under FAA and EASA and TCCA reg. frameworks)

**Numerical Notes:**
- [1] - Production Manufacturing Approval (PMA) parts may only be accepted as detailed in EASA Part-21 or EU-US bilateral agreement/TIP (e.g. critical components not designed by the TC holder are excluded)
- [2] - Product Design Approval (PDA) parts may only be accepted as detailed in EASA Part-21 or Canada-US bilateral agreement/TIP (e.g. critical components not designed by the TC holder are excluded)
- [3] - Approval of Approved Aeronautical Product (AAPA) parts may only be accepted as detailed in EASA Part-21 or Brazil-US bilateral agreement/TIP (e.g. critical components not designed by the TC holder are excluded)
- [4] - EASA Form 1 is the only acceptable release document for used parts.

### Definitions:
- A component referred to in 21.A.307(c) of Annex I (Part-21) to Reg. (EU) 748/2012 for ELA1 or ELA2 aircraft) is not subject to this FAQ; they shall only be accepted if considered eligible for installation by the aircraft owner in its own aircraft.
- Standard parts are not subject to this FAQ; they shall be accompanied by an evidence of conformity (e.g. Certificate of Conformity (CoC)) traceable to the applicable standard (see AMC M.A.501(c) to Annex I (Part-M) to Reg. (EU) 1321/2014).

### Notes cont.
- FAA Form 8130-3 containing FAA and EASA release.
- Not accepted
- FAA Form 8130-3 [not containing EASA release] or FAA Form 8130-3 DUAL [EASA & FAA]
- In general not accepted, specific cases are listed on EASA website.

### References:

### Convention used in the table:
- Used component:
  - Reg. authority Form SINGLE/DUAL/TRIPLE (Reg. framework(s) under which the component maintenance is released) Form has been issued under a maintenance approval, i.e. signed on the right hand side of the document.
  - Example: FAA Form 8130-3 DUAL [EASA & FAA] means a Form 8130-3 containing FAA and EASA release.
- New component:
  - Reg. authority Form NEW Form has been issued under a production approval, i.e. signed on the left hand side of the document.

### Notes cont.
- JAA Form One issued prior to 28 November 2004 by a JAR 145 organisation approved by a JAA Full Member State is considered equivalent to an EASA Form 1 (used component)
- JAA Form One issued prior to 28 September 2001 by a production organisation approved by a competent authority in accordance with its national regulations is considered equivalent to an EASA Form 1 (new component)
- FAA Form 8130-3 stating compliance with EU Regulation only and issued by organisation located in the US may be acceptable under specific condition expressed in the EU - USA Bilateral agreement.
- For the specific case of rebuilt engine/component, please refer to FAQ n.43771 [FAQ n.43771](https://www.easa.europa.eu/document-library/bilateral-agreements/eu-usa)

### Caution:
- Components referred to in 21.A.307(c) of Annex I (Part-21) to Reg. (EU) 748/2012 (for ELA1 or ELA2 aircraft) are not subject to this FAQ; they shall only be accepted if considered eligible for installation by the aircraft owner in its own aircraft.
- Standard parts are not subject to this FAQ; they shall be accompanied by an evidence of conformity (e.g. Certificate of Conformity (CoC)) traceable to the applicable standard (see AMC M.A.501(c) to Annex I (Part-M) to Reg. (EU) 1321/2014)

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