

Certification Directorate
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Report

EU/US BASA Supporting SSD List for CS 23



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1 Executive Summary

This report provides the list of Significant Standard Difference (SSDs) between the various CS23 amendment levels and the corresponding (by applicability date) 14CFR FAR23 amendment levels

These lists are provided to support the implementation of TIP Revision 6 of the EU/US Bilateral for products certified under the JAR/CS23 or FAR 23 Airworthiness codes.

The lists identifies the differences that exist between the comparison of CS 23 and FAR 23 for each amendment pair as a list of Significant Standard Differences. These are requirement pairs where the applicant is required to provide additional information in support of the EASA validation process to ensure compliance has been demonstrated to the EASA Certification Basis as the Validating Authority, in addition to the compliance demonstration for the FAA Certification Basis, as the Certifying Authority.

This document does not currently cover the Amendment Par CS23 Amendment 4 and FAR23 Amendment 64. This document will be updated at a later date to include this comparison.

2 Significant Standard Differences

The following tables identify the list of Significant Standard Differences that need to be considered when conducting an EASA validation of a US FAA certification approval. The lists are presented for each amendment pair of CS/FAR 23.

The tables are presented in Amendment Pairs based on the chronological issue date of each Amendment. The method for use is to establish the reference application date, as per the requirements of Part 21 and the provisions in the TIP. The table below can then be used to identify the applicable amendment pair.

Once the Amendment Pair is established then the tables under Section 2.1 to 2.12 can be utilised to identify the appropriate SSD lists based on the FAA Certification Basis. Please note the list are compiled based on the changes that were introduced with each Amendment revision and therefore need to be treated as a compound list to ensure full identification of the SSDs for each amendment pair.

For example with an Amendment Pair of CS23 Amendment 1 FAR23 Amendment 59 (Section 2.5) all tables in Sections 2.1 to 2.5 must be reviewed for the applicable SSD list.

Release Date	CS 23 Amendment Level	FAR 23 Amendment Level
14 th November 2003	Amendment 0	Up to and Including Amendment 55
5 th September 2007	Amendment 0	Amendment 57 (no Amendment 56)
4 th April 2008	Amendment 0	Amendment 58
23 rd December 2008	Amendment 0	Amendment 59
12 th February 2009	Amendment 1	Amendment 59
9 th July 2009	Amendment 1	Amendment 58 (Corrected)
28 th September 2010	Amendment 2	Amendment 58 (Corrected)
8 th August 2011	Amendment 2	Amendment 61 (no amendment 60)
31 st January 2012	Amendment 2	Amendment 62
20 th July 2012	Amendment 3	Amendment 62
15 th July 2015	Amendment 4	Amendment 62
21 st March 2017	Amendment 4	Amendment 63

The comparison lists does not currently include the latest Amendment pair with the redefinition of CS/FAR23 around the ASTM standards. This document will be updated at a later date to include this Amendment pair.

2.1 CS 23 Amendment 0/ FAR 23 up to Amendment 55

CS 23 Amendment 0 was issued on the 14th November 2003, this corresponds with FAR 23 Amendment 55, issued on the 1st March 2002. Before the establishment of EASA, on the 28th September 2003, member states used a variety of airworthiness standards (JAR, BACR, etc) no standard differences have been established for these codes with FAR 23.

Item	CS 23 Requirement	Detail
SUBPART A- GENERAL		
1	23.1	Applicability No SSD exists if compliance to FAR 23.3 is demonstrated except for Commuter Category aeroplanes, CS 23 limits commuter aeroplanes to twin-engine propeller driven, FAR 23 allows for multi-engine propeller driven. An SSD exists only for aeroplanes that have more than 2 propeller driven engines.
SUBPART B-FLIGHT:		PERFORMANCE
2	23.49	Stalling Speed CS23.49(c) SSD exists if compliance is demonstrated to FAR23.49(d), CS 23 contains no such alleviation for stalling speed V_{SO} greater than 113 km/h (61 knots) for single-engine or twin-engine below 2722 Kg (6000lb). CS23.49(c)(2) SSD exists only for aeroplanes that have more than 2 engines.
3	23.51	Take-off Speeds CS23.51(a)(1) SSD exists only for aeroplanes that have more than 2 engines. CS23.51(b)(1) SSD exists only for aeroplanes that have more than 2 engines.
4	23.67	Climb: One-Engine-Inoperative CS23.67(a)(1)&(2)no CS alleviation for aeroplanes that meet the FAR23.562(d)
SUBPART B-FLIGHT:		CONTROLLABILITY AND MANOEUVRABILITY
5	23.145	Longitudinal Control CS23.145(e)(1)&(2) SSD exists only for aeroplanes that have more than 2 engines.
6	23.147	Directional and lateral control CS23.147(a) SSD exists only for aeroplanes that have more than 2 engines. CS23.147(b) SSD exists only for aeroplanes that have more than 2 engines.
7	23.157	Rate of roll CS23.157(c)(2) requires a maximum 7 second hard limit
SUBPART B-FLIGHT:		TRIM
8	23.161	Trim CS23.161(d) SSD exists only for aeroplanes that have more than 2 engines.
SUBPART B-FLIGHT:		SPINNING
9	23.221	Spinning CS23.221(a) does not allow for compliance to the FAR optional spin resistant requirements of FAR23.221.(a)(2).
SUBPART C-STRUCTURE:		VERTICAL SURFACES
10	23.443	<i>Reserved</i>
SUBPART C-STRUCTURE:		GROUND LOADS
11	23.473	Ground Load Conditions and Assumptions CS23.473(c) SSD exists only for aeroplanes that have more than 2 engines.
SUBPART C-STRUCTURE:		WATER LOADS

Item	CS 23 Requirement	Detail
12	23.527	<i>Reserved</i>
13	23.535	<i>Reserved</i>
SUBPART C-STRUCTURE:		EMERGENCY LANDING CONDITIONS
14	23.562	Emergency Landing Dynamic Conditions CS23.562(b) applies to Commuter aeroplanes, FAR 23.562(b) is not applicable to Commuter FAR23.562(d) use of this paragraph is not permitted under CS23.562 (see item 2 above)
SUBPART D-DESIGN AND CONSTRUCTION:		CONTROL SYSTEMS
15	23.677	Trim Systems CS23.677 SSD exists only for aeroplanes that have more than 2 engines.
16	23.701	Flap Interconnection CS23.701(c) SSD exists only for aeroplanes that have more than 2 engines.
SUBPART D-DESIGN AND CONSTRUCTION:		PERSONNEL AND CARGO ACCOMMODATIONS
17	23.777	Cockpit Controls CS23.777(c)(1) SSD exists only for aeroplanes that have more than 2 engines. CS23.777(e)(1) SSD exists only for aeroplanes that have more than 2 engines.
18	23.785	Seats, Berths, Litters, Safety Belts and Shoulder Harnesses CS23.785(c) reference to the inertia loads generated for Commuter aeroplanes under CS23.562, which is not applicable for FAR23 Commuter aeroplanes
19	23.807	Emergency Exits CS23.807(e) SSD exists only for aeroplanes that have more than 2 engines.
20	23.815	Width of Aisle CS23.815(b) SSD exists only for aeroplanes with exactly 10 seats due to error in CS23.
SUBPART E-POWERPLANT		GENERAL
21	23.901	Installation 23.901(d)(2) CS 23 sets minimum requirements for water ingestion
22	23.903	Engines and Auxiliary Power Units CS23.903 SSD exists for installed APUs CS23.903(a)(2) SSD exists if compliance has been demonstrated to FAR23.903(a)(2)(i), (ii), (iii), no SSD if FAR23.903(a)(2)(iv) is complied with. CS23.903(c) CS23.903 does not allow a single fuel tank for twin engine installations.
23	23.904	Automatic Power Reserve System 23.904 SSD exists only for aeroplanes that have more than 2 engines.
24	23.925	Propeller Clearance 23.925 CS more prescriptive FAR allows clearance reductions with substantiation CS does not

Item	CS 23 Requirement	Detail
25	23.929	Engine Installation Ice Protection 23.929 SSD only exists for wooden propellers.
26	23.933	Turbojet and Turbofan Engine Thrust Reverser System Tests 23.933(b)(3) Commuter category requirement in CS no FAR equivalent.
27	23.934	Turbojet and turbofan engine thrust reverser system tests CS refers to CS E requirements, FAR references FAR-33.
SUBPART E-POWERPLANT		FUEL SYSTEM
28	23.953	Fuel System Independence 23.953(a) does not allow a single fuel tank for twin engine installations, no alleviations granted under FAR23.953(b) are accepted by CS23.953. In addition SSD exists for aeroplanes that have more than 2 engines.
29	23.955	Fuel Flow 23.955(a) CS applies to all engine types and “engine operation” FAR23.955(a) limited to “carburettor operation” only. 23.955(f)(2) SSD exists only for aeroplanes that have more than 2 engines.
30	23.967	Fuel Tank Installation 23.967(e)(3) No FAR requirement.
SUBPART E-POWERPLANT		FUEL SYSTEM COMPONENTS
31	23.993	Fuel Systems Lines and Fittings 23.993(e) SSD exists for installed APUs
SUBPART E-POWERPLANT		OIL SYSTEM
32	23.1011	General 23.1011(b) SSD exists for installed APUs
SUBPART E-POWERPLANT		COOLING
33	23.1047	Cooling 23.1047 SSD exists only for aeroplanes that have more than 2 engines.
SUBPART E-POWERPLANT		INDUCTION SYSTEM
34	23.1091	Induction System Icing Protection 23.1091(c)(1) SSD exists for installed APUs
35	23.1093	Induction System Icing Protection 23.1093(a)(5) CS requires a higher temperature rise, in addition SSD exists for aeroplanes that have more than 2 engines.
SUBPART E-POWERPLANT		POWERPLANT FIRE PROTECTION
36	23.1189	Shut-Off Means 23.1189(a) SSD exists only for aeroplanes that have more than 2 engines. 23.1189(a)(5) CS specifies “hazardous amount” FAR sets a limit of “one quart”
37	23.1193	Cowling and Nacelle 23.1193(f) SSD exists only for aeroplanes that have more than 2 engines.
38	23.1195	Fire Extinguishing Systems CS 23.1195(a)(1) CS requires “fire zone” FAR requires “engine compartment”

Item	CS 23 Requirement	Detail
39	23.1203	Fire Detector System 23.1203(a)(1) CS requires "fire zone" FAR requires "Engine Compartment". 23.1303(a)(1)(i) & (ii) SSD exists only for aeroplanes that have more than 2 engines.
SUBPART F-EQUIPMENT		GENERAL
40	23.1305	Powerplant Instruments 23.1305(b)(4) CS requires fuel pressure indicator only FAR allows other options.
41	23.1309	Equipment, Systems and Installations 23.1309(d) SSD exists only for aeroplanes that have more than 2 engines.
SUBPART F-EQUIPMENT		INSTRUMENTS: INSTALLATION
42	23.1321	Arrangement and Visibility 23.1321(b) SSD exists only for aeroplanes that have more than 2 engines.
43	23.1325	Static Pressure System CS 23.1325(g) reference to CS 23.1525(b)(3) FAR reference 23.1559(b).
SUBPART F-EQUIPMENT		ELECTRICAL SYSTEMS AND EQUIPMENT
44	23.1351	General CS 23.1351(f) requires consideration to: "The external power connection must be located so that its use will not result in a hazard to the aeroplane or ground personnel."
45	23.1365	Electric Cables and Equipment 23.1365(e)(1) & (2) No corresponding FAR requirement
46	23.1389	Position Light Distribution and Intensities CS 23.1389(b) additional reference to "horizontal plane minimum intensity"
SUBPART F-EQUIPMENT		MISCELLANEOUS EQUIPMENT
47	23.1437	Accessories for Twin-Engine Aeroplanes 23.1437 SSD exists only for aeroplanes that have more than 2 engines.
SUBPART G- OPERATING LIMITATIONS AND INFORMATION		MARKINGS AND PLACARDS
48	23.1545	Airspeed Indicator. 23.1545(a) SSD exists for mechanical instruments only. 23.1545(b)(5)&(6) SSD exists only for aeroplanes that have more than 2 engines.
SUBPART G- OPERATING LIMITATIONS AND INFORMATION		AEROPLANE FLIGHT MANUAL
49	23.1585	Operating Procedures 23.1585(e) & (g) SSD exists only for aeroplanes that have more than 2 engines.
50	23.1587	Performance Information 23.1587(c)(4)&(5) SSD exists only for aeroplanes that have more than 2 engines.

2.2 CS 23 Amendment 0/ FAR 23 Amendment 57

With the publication of FAR23 Amendment 57 (there was no Amendment 56) on the 5th September 2007 no new SSDs were identified with CS23 Amendment 0.

2.3 CS 23 Amendment 0/ FAR 23 Amendment 58

With the publication of FAR23 Amendment 58 on the 4th April 2008 the following new SSDs were identified with CS23 Amendment 0.

Item	CS 23 Requirement	Detail
SUBPART F-EQUIPMENT		MISCELLANEOUS EQUIPMENT
51	23.1457	Cockpit Voice Recorders SSD exists when compliance is shown to FAR23.(e)(2).

2.4 CS 23 Amendment 0/ FAR 23 Amendment 59

With the publication of FAR23 Amendment 59 on the 23rd December 2008 the following new SSDs were identified with CS23 Amendment 0.

Item	CS 23 Requirement	Detail
SUBPART E-POWERPLANT		GENERAL
52	23.907	Propeller Vibration 23.907(a)(1) CS23 does not allow for analysis

2.5 CS 23 Amendment 1/ FAR 23 Amendment 59

With the publication of CS23 Amendment 1 on the 12th February 2009 the following new SSDs were identified with FAR23 Amendment 59.

Item	CS 23 Requirement	Detail
SUBPART B-FLIGHT:		PERFORMANCE
2	23.49	Stalling Speed CS23.49(c) SSD exists if compliance is demonstrated to FAR23.49(d), CS 23 contains no such alleviation for stalling speed V_{SO} greater than 113 km/h (61 knots) for single engine or twin engine below 2722 Kg (6000lb). SSD Deleted
53	23.49	Stalling Speed CS23.49(d) SSD exists only for aeroplanes that have more than 2 engines.
SUBPART C-STRUCTURE:		EMERGENCY LANDING CONDITIONS
14	23.562	Emergency Landing Dynamic Conditions CS23.562(b) applies to Commuter aeroplanes, FAR 23.562(b) is not applicable to Commuter FAR23.562(d) use of this paragraph is not permitted under CS23.562 (see item 2 above) SSD Deleted

2.6 CS 23 Amendment 1/ FAR 23 Amendment 58 (Corrected)

With the publication of FAR23 Amendment 58 (Corrected) on the 9th July 2009 no new SSDs were identified with CS23 Amendment 1.

2.7 CS 23 Amendment 2/ FAR 23 Amendment 58 (Corrected)

With the publication of CS23 Amendment 2 on the 28th September 2010 no new SSDs were identified with FAR23 Amendment 58 (Corrected).

2.8 CS 23 Amendment 2/ FAR 23 Amendment 61

With the publication of FAR23 Amendment 61 (there was no amendment 60) on the 8th August 2011 no new SSDs were identified with CS23 Amendment 2.

2.9 CS 23 Amendment 2/ FAR 23 Amendment 62

With the publication of FAR23 Amendment 62 on the 31st January 2012 the following new SSDs were identified with CS23 Amendment 2.

Item	CS 23 Requirement	Detail
SUBPART A- GENERAL		
54	23.1	<p>Applicability No SSD exists if compliance to FAR 23.3 is demonstrated except for Commuter Category aeroplanes, CS 23 limits commuter aeroplanes to twin-engine propeller driven, FAR 23 @62 allows for multi-engine, propeller or turbine driven. An SSD exists only for aeroplanes that have either 2, or more, Turbojet engines or more than 2 propeller driven engines.</p> <p>Note: Unless notified below any SSD that has been established above for Commuter aeroplanes with more than 2 engines is applicable to Commuter aeroplanes with 2, or more, Turbojet engines.</p>
SUBPART B-FLIGHT: PERFORMANCE		
55	23.45	<p>General CS23.45(h) SSD exists for other than Commuter aeroplanes.</p>
56	23.51	<p>Take-off Speeds CS23.51(c) SSD exists for other than Commuter aeroplanes.</p>
57	23.53	<p>Take-off Performance CS23.53(c) SSD exists for other than Commuter aeroplanes.</p>
58	23.55	<p>Accelerate-Stop Distance CS23.55 SSD exists for other than Commuter aeroplanes.</p>
59	23.57	<p>Take-off Path CS23.57 SSD exists for other than Commuter aeroplanes.</p>

Item	CS 23 Requirement	Detail
60	23.59	Take-off Distance and Take-off Run CS23.59 SSD exists for other than Commuter aeroplanes.
61	23.61	Take-off flight Path CS23.61 SSD exists for other than Commuter aeroplanes.
62	23.63	Climb: General CS23.63(c) SSD exists for single-engine turbines, and multiengine turbine airplanes of 6,000 pounds or less maximum weight in the normal, utility, and acrobatic category. CS23.63(d) SSD exists for other than Commuter aeroplanes.
63	23.67	Climb: One-Engine-Inoperative CS23.67 SSD exists for all aeroplane types due to the changes in aeroplane categorisation.
64	23.73	Reference Landing Approach Speed 23.73(b) SSD exists for normal, utility, and acrobatic category turbine powered airplanes of 6,000 pounds or less maximum weight, turboprops of more than 6,000 pounds maximum weight 23.73(c) SSD exists for other than Commuter aeroplanes.
65	23.77	Balked Landing 23.77(b) SSD exists for normal, utility, and acrobatic category turbine powered airplanes of 6,000 pounds or less maximum weight, turboprops of more than 6,000 pounds maximum weight. 23.77(c) SSD exists for other than Commuter aeroplanes.
SUBPART B-FLIGHT:		STABILITY
66	23.181	Dynamic Stability CS23.181(b) does not allow for great than 7 cycles to damp.
SUBPART B-FLIGHT:		STALLS
66	23.201	Wings Level Stall CS23.201(d) & (e) does not allow for the alleviation above 25,000ft
67	23.203	Turning Flight and Accelerated Turning Stalls CS23.203(c) SSD for Turbine powered aeroplanes
SUBPART D-DESIGN AND CONSTRUCTION:		PERSONNEL AND CARGO ACCOMMODATIONS
68	23.777	Cockpit Controls CS23.777(c)(4) requires additional prescriptive dimensions
69	23.807	Emergency Exits CS23.807(e) does not allow the alleviation presented by FAR23.807(e)(3).
SUBPART D-DESIGN AND CONSTRUCTION:		PRESSURATION
70	23.841	Cockpit Controls CS23 limits cabin altitude to not more than 15,000ft in the event of any probable failure or malfunction.
SUBPART D-DESIGN AND CONSTRUCTION:		FIRE PROTECTION
70	23.853	Passenger and Crew Compartment Interiors CS23.853(d)(2) has additional requirements not present in FAR23.853(d)(2).

Item	CS 23 Requirement	Detail
SUBPART E-POWERPLANT		POWERPLANT CONTROLS AND ACCESSORIES
71	23.1165	Engine Ignition Systems 23.1165(f) SSD for Turbine powered aeroplanes
SUBPART E-POWERPLANT		POWERPLANT FIRE PROTECTION
71	23.1189	Shut-Off Means 23.1189(a) SSD exists only for aeroplanes that have more than 2 engines. 23.1189(a)(5) CS specifies "hazardous amount" FAR sets a limit of "one quart"
SUBPART E-POWERPLANT		POWERPLANT FIRE PROTECTION
72	23.1193	Cowling and Nacelle CS23.1193(g) SSD exists for Commuter aeroplanes as not limited by aeroplane/engine configuration.
73	23.1195	Fire Extinguishing Systems CS23.1195(a) SSD exists for Commuter aeroplanes as not limited by aeroplane/engine configuration.
74	23.1197	Fire Extinguishing Agents CS23.1197 SSD exists for Commuter aeroplanes as not limited by aeroplane/engine configuration.
75	23.1199	Extinguishing Agent Containers CS23.1199 SSD exists for Commuter aeroplanes as not limited by aeroplane/engine configuration.
76	23.1201	Fire Extinguishing System Materials CS23.1201 SSD exists for Commuter aeroplanes as not limited by aeroplane/engine configuration.
SUBPART F - EQUIPMENT		GENERAL
76	23.1303	Function and Installation CS23.1303(c) SSD exists when not using a non-stabilised magnetic direction indicator. CS23.1303(e) SSD exists for Turbine engine-powered aeroplanes
77	23.1309	Equipment, Systems and Installations SSD exists as CS is more prescriptive.
SUBPART F - EQUIPMENT		INSTRUMENTS: INSTALLATION
78	23.1331	Instruments Using a Power Source CS23.1331(c) SSD exists for aeroplanes not certified to IFR.
SUBPART G- OPERATING LIMITATIONS AND INFORMATION		MARKINGS AND PLACARDS
48	23.1545	Airspeed Indicator 23.1545(a) SSD exists for mechanical instruments only. SSD Deleted 23.1545(b)(5)&(6) SSD exists only for aeroplanes that have more than 2 engines.
79	23.1555	Control Markings CS23.1551(d) SSD exists when complying with FAR23.1555(d)(3) as no alleviation allowed.

2.10 CS 23 Amendment 3/ FAR 23 Amendment 62

With the publication of CS23 Amendment 3 on the 20th July 2012 no new SSDs were identified with FAR23 Amendment 62.

2.11 CS 23 Amendment 4/ FAR 23 Amendment 62

With the publication of CS23 Amendment 4 on the 15th July 2015 no new SSDs were identified with FAR23 Amendment 62.

2.12 CS 23 Amendment 4/ FAR 23 Amendment 63

With the publication of FAR23 Amendment 63 on the 21st March 2017 no new SSDs were identified with CS23 Amendment 4.

2.13 Other CS Requirements that may be required at a product level.

The following lists other CS requirements that have no direct equivalent FAR requirements therefore if required at a product level compliance demonstration by the applicant.

No standard differences have been established for these codes with 14 CFR FAR. If an applicant requires these comparisons they must contact EASA for guidance.

	CS Requirement	Detail
1	CS-ACNS	No FAR Requirement to compile SSD list
2	CS-FCD	No FAR Requirement to compile SSD list
3	CS-MCSD	No FAR Requirement to compile SSD list
4	CS-MMEL	No FAR Requirement to compile SSD list
5	CS-GEN-MMEL	No FAR Requirement to compile SSD list
6	CS-SIMD	No FAR Requirement to compile SSD list
7	CS-APU	No FAR Requirement to compile SSD list
8	CS-AWO	No FAR Requirement to compile SSD list
9	CS-CO2	No FAR Requirement to compile SSD list