

EUROPEAN OPERATORS FLIGHT DATA MONITORING FORUM (EOFDM)  
WORKING GROUP C

SAFETY PROMOTION

Good practice document

# FLIGHT DATA MONITORING ANALYSIS TECHNIQUES AND PRINCIPLES

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## Note

### About the authors

This document was produced by the working group C of the European Operators Flight Data Monitoring forum (EOFDM WG-C – Integration of an FDM programme into operator processes). Unless otherwise indicated, all figures shown in this document were provided by members of working group C.

According to its terms of reference, the EOFDM is a voluntary and independent safety initiative. Therefore, this document should be considered as industry good practice which EASA promotes actively. This document should not be considered as an alternative to any applicable regulatory requirement, and should not be considered as official guidance from EASA.

Information on the EOFDM forum and other good practice documents produced by EOFDM can be consulted on the EASA website (<https://www.easa.europa.eu/>).

### Whom is this document for?

This document is primarily addressed to safety managers and flight data monitoring (FDM) specialists working for organisations (air operators) that already run an FDM programme. Therefore, it does not contain an introduction to FDM or guidance on performing the safety risk assessment.

Guidance and examples provided in this document are only meant to support the thinking process of FDM specialists and the enhancement of their FDM programme. They are just indicative and not exhaustive.

### FDM data protection and confidentiality

The integrity of FDM programmes rests upon protection of the FDM data. Any disclosure of data that reveals flight crew member identity for purposes other than safety management can compromise the voluntary provision of safety data, thereby compromising flight safety. ICAO standards and recommended practices on flight data monitoring are in ICAO Annex 6 Part I (aeroplanes) 3.3, and Part III (helicopters) Section II, 1.3.

The data access and security policy should restrict access to FDM information to authorised persons. In addition, when data access is required for airworthiness and maintenance purposes, a procedure should be in place to prevent disclosure of flight crew member identity. This procedure should be written in a document, which should be signed by all parties (detailed specifications regarding such procedure can be found in Part-ORO of the EU rules for air operations<sup>1</sup>, AMC1 ORO.AOC.130).

### Feedback on this document

In case you want to comment or give your feedback on this document, please contact: [fdm@easa.europa.eu](mailto:fdm@easa.europa.eu).

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<sup>1</sup> The annexes to Commission Regulation (EU) 965/2012 contain the EU rules for air operations. A consolidated version of these rules can be consulted at: <https://www.easa.europa.eu/document-library/easy-access-rules/easy-access-rules-air-operations>

## Executive summary

This document provides industry good practice regarding common analysis techniques used by flight data monitoring specialists. It also offers some principles to be aware of for successful implementation of these analysis techniques.

Chapter 1 contains advice for identifying safety issues that can be monitored with the help of FDM and for scoping the FDM algorithms.

Chapter 2 contains an overview of the design, testing, production, and maintenance phases of an FDM algorithm. It also provides the reader with experience and practical guidance for each of these phases.

Chapter 3 is focussed on the use of statistics for analysing the output of FDM algorithms and presenting the results of such analyses. Chapter 3 only introduces some statistical notions and tools that are interesting for analysing flight data and it does not include advanced statistics.

## Abbreviations and symbols

ADS-B	automatic dependent surveillance-broadcast
AFM	aircraft flight manual
AMM	aircraft maintenance manual
ATQP	alternative training & qualification programme
ARINC	Aeronautical Radio, Incorporated
AMC	acceptable means of compliance
BI	business intelligence
CAST	Commercial Aviation Safety Team
CAT	commercial air transport
CFIT	controlled flight into terrain
CRM	crew resource management
FATO	final approach and take-off area
FDM	flight data monitoring
FDR	flight data recorder
ICAO	International Civil Aviation Organization
LOC-I	loss of control in flight
MAC	mid-air collision
MOR	mandatory occurrence report
MEL	minimum equipment list
ORO	organisation requirements for air operations
RE	runway excursion
SD	standard deviation
SID	standard instrument departure
SMS	safety management system
SOP(s)	standard operating procedure(s)
TAWS	terrain awareness & warning system
VOR	voluntary occurrence report

## Definitions

<b>Data snapshot (of an FDM event)</b>	Data automatically extracted, processed, and recorded when defined conditions are detected.
<b>Flight data</b>	Parametric data collected on-board the aircraft by a system dedicated for this purpose (for instance, a (wireless) quick-access recorder).
<b>FDM algorithm</b>	An FDM event algorithm or an FDM measurement algorithm.
<b>FDM data</b>	Flight data collected and data produced in the framework of an FDM programme.
<b>FDM event</b>	An individual detection of defined conditions, either created by an FDM event algorithm, or obtained by applying an algorithm to detect such conditions on an FDM measurement.
<b>FDM event algorithm</b>	Algorithm processing flight data to detect defined conditions (trigger logic) and extract data whenever these defined conditions are met (data snapshot).
<b>FDM measurement</b>	Data created by applying an FDM measurement algorithm on the flight data of a batch of flights.
<b>FDM measurement algorithm</b>	Algorithm processing flight data to extract data meeting certain defined conditions from each flight.
<b>FDM programme</b>	Programme for routine collection and analysis of flight data to develop objective and predictive information for advancing safety, e.g. through improvements in flight crew performance, training effectiveness, operational procedures <sup>2</sup> .
<b>Hazard</b>	A condition or an object with the potential to cause or contribute to an aircraft incident or accident <sup>3</sup> .
<b>Safety risk</b>	The predicted probability and severity of the consequences or outcomes of a hazard <sup>4</sup> .
<b>Threshold or trigger value</b>	Constant or condition-based value included in an algorithm, and beyond which an event is detected (e.g. by an FDM event algorithm) or an action is due.
<b>Trigger logic (of an FDM event algorithm)</b>	Part of the FDM event algorithm that contains the conditions triggering detection.

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<sup>2</sup> According to the definition of FDM in Annex I to Regulation (EU) 965/2012 (Definitions applicable to the rules for air operation), it is a non-punitive use of flight data. EOFDM document 'Breaking the silos' provides guidance on Just Culture and reconciling the non-punitive character of FDM with the identification of unacceptable behaviour.

<sup>3</sup> Definition provided in ICAO doc 9859 (Safety Management Manual).

<sup>4</sup> Definition provided in ICAO doc 9859.

## I. Identifying the relevant FDM algorithms

### *Note:*

The 'FDM specialist' may be an internal resource (employed by the operator) or external resource (if part of the FDM programme is subcontracted to a third party).

### 1. Identifying and prioritising issues to be monitored through FDM

#### FDM for safety performance monitoring and hazard identification

The FDM programme is an essential component of the SMS<sup>5</sup> both in terms of hazard identification and safety performance monitoring and measurement<sup>6</sup> and operators often summarise their main operational safety risks in a so-called 'safety risk register'<sup>7</sup>. While the scope of the FDM programme should be consistent with the risk register, it is advisable to not restrict the scope of work of the FDM specialist to just transposing the safety risk register into FDM events and measurement algorithms.

Just like any other member of the safety manager's team, the FDM specialist should be allowed to pursue new analyses if he/she suspects or identifies a new threat for the safety of its operations. FDM is not only a tool for monitoring known safety risks but should also be used to identify new hazards and detect safety threats unknown or not quantifiable through other data sources.

#### Identifying different types of safety risks

Safety risk can be categorised as operational or non-operational. Operational safety risks are risks associated to hazards that occur when the aircraft is operated and that may act as a precursor to an incident or accident (e.g. icing conditions, bird strike, proximity with terrain, etc.). Non-operational safety risks are related to other kinds of factors, such as organisational, related to training, company safety culture, etc. and are mostly considered as contributing factors.

In practice, the FDM specialist will mostly focus on operational safety risks. This is because FDM measures the actual operation of the aircraft. However, in some cases trends and events can be used to find cues of non-operational issues, such as inadequate training (e.g. wrong take-off technique or response to TCAS RA) or culture (e.g. non-adherence to stabilised approach criteria). In such cases it is essential that the FDM specialists bring their findings to the attention of the safety manager and that they can be combined with other data sources (e.g. occurrence reports, interviews, surveys, weather & ADS-B data, etc.) to investigate the operational context and possible causes. Vice versa, FDM can be used to assess threats identified by other means e.g. incident information from an internal crew report or external safety publication ('Does this happen in our company?').

#### Need for additional data sources

Although FDM provides a wealth of data, it provides a limited understanding of the operation. Other factors, such as weather conditions (e.g. visibility, the position of advective weather), factors affecting

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<sup>5</sup> According to the EU rules for air operations, Part-ORO, ORO.GEN.200 an operator shall identify, evaluate, and manage the risks entailed by its activities, including taking actions to mitigate the risk and verify their effectiveness.

<sup>6</sup> According to ICAO Annex 19, Appendix 2, Safety performance monitoring and measurement is one of the twelve elements that are part of SMS implementation.

<sup>7</sup> A safety risk register is a tool to manage safety risks. How to establish a safety risk register is out of the scope of this document. Refer to ICAO doc 9859, Section 2.5.

the individual and collective performance of flight crew members, distractions and disturbances, workload, traffic situation, frequency congestion, etc. must be taken into account to build a more complete picture. Therefore, it is important not to draw conclusions solely based on FDM data. Other data should be consulted to understand the actions, inactions, or decisions which led to the operation of the aircraft as logged by the on-board sensors. When identifying which issues to monitor, the availability of necessary data sources should be taken into account.

### Sources to identify risks

In addition to the safety risk register, the occurrences listed in Annex I to Commission Implementing Regulation (EU) 2015/1018<sup>8</sup> (Occurrences related to the operation of the aircraft) may be considered, as FDM algorithms related to some of these occurrences could be programmed.

EOFDM documents titled ‘Review of FDM precursors’ and ‘Guidance for the implementation of FDM precursors’ cover the risks of runway excursion (RE), loss of control in flight (LOC-I), controlled flight into terrain (CFIT) and mid-air collision (MAC) for aeroplanes and it is advised to consult these documents<sup>9</sup>. However, it is not expected that an operator implements all the precursors proposed in those documents. Guidance on establishing a relevant set of FDM algorithms (FDM event algorithms and FDM measurement algorithms) is provided in this chapter.

With regards to helicopter offshore operations, Part-SPA of the EU rules for air operations<sup>10</sup>, GM2 SPA.HOFO.145, and HeliOffshore HFDM Recommended Practice<sup>11</sup> contain examples of FDM events.

### Describing an operational safety risk

In order to define a set of FDM algorithms related to a given operational safety risk, the FDM specialist should be provided with an accurate and complete description of the related hazard. This implies the following:

1. The safety manager team, including the FDM specialist should use a common taxonomy<sup>12</sup> to describe the occurrence category and flight phase.
2. The description of each operational safety risk should include, as a minimum:
  - the flight phase(s) where the hazard may occur or is assumed to occur;
  - the concerned inbound/outbound route, airport, and runways;
  - environmental context (e.g. time of day/night, day of the week, season, etc.);
  - the affected aircraft model(s) and, if applicable, the concerned airfield; and
  - any additional factor that increases the probability of occurrence of this risk.

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<sup>8</sup> Commission Implementing Regulation (EU) 2015/1018 of 29 June 2015 laying down a list classifying occurrences in civil aviation to be mandatorily reported according to Regulation (EU) No 376/2014 of the European Parliament and of the Council.

<sup>9</sup> These documents are published on EASA website (see <https://www.easa.europa.eu/domains/safety-management/safety-promotion/european-operators-flight-data-monitoring-eofdm-forum> )

<sup>10</sup> The annexes to Commission Regulation (EU) 965/2012 contains the EU rules for air operations. A consolidated version of these rules can be consulted at: <https://www.easa.europa.eu/document-library/easy-access-rules/easy-access-rules-air-operations>

<sup>11</sup> <http://heli-offshore.org/wp-content/uploads/2021/01/HFDM-RP-v1.0-1.pdf>

<sup>12</sup> For example, the CAST/ICAO common taxonomy team publishes common taxonomies and definitions for aviation accident and incident reporting systems (see <http://www.intlaviationstandards.org> )

Example: ‘risk of runway overrun when taking off from a limitative runway with aircraft model X’ instead of just ‘risk of runway excursion’.

3. The description of the operational safety risk should be refined by:

- an analysis of internal incident and hazard reports;
- research of published accident investigation reports or other guidance material relevant to that safety risk; and
- a clear understanding of the operational reality (work-as-done) for example, by including a practical explanation from a pilot on possible errors and their consequence.

Example: wrong setting of the horizontal stabiliser (trim) may translate into an abnormal low pitch rate at lift-off or unusual and very strong inputs on flight controls to get the aircraft airborne.

The FDM specialist may have to engage with the staff documenting the safety risks for the SMS, to refine the description of an operational safety risk, until this description is clear and detailed enough. This usually is an iterative process. This process should be documented for ensuring that the programming of FDM algorithms is based on commonly understood needs. This documentation should be retained as part of the SMS documentation of the operator.

The FDM specialist should have access to detailed description of standard operating procedures (SOPs) and limitations to program the actual event. Refer to Chapter II.1 Defining an FDM algorithm.

## 2. Understanding the possibilities and limitations of the FDM programme

When is FDM relevant for the monitoring of an operational safety risk?

Flight data can provide accurate information about aircraft trajectory parameters, status of aircraft systems, and human-machine interactions. However, flight data recording systems have very limited capability to record information external to the aircraft such as other traffic, ATC communications and weather (e.g. visibility, light conditions), human factors (e.g. crew resource management, fatigue) or damage to the aircraft structure. Thus, it is advisable, where possible, to combine flight data with other operational data (e.g., flight number, timestamp, etc.).

Is there a reliable system for collecting flight data for the FDM programme?

A flight collection rate of 80 % or more is advisable for making the FDM programme effective<sup>13</sup> and it is a condition for implementing an Alternative Training & Qualification Programme (ATQP)<sup>14</sup>. Wireless transmission of flight data to the ground may be advantageous with this regard, as it avoids moving recording media between the aircraft and maintenance facilities.

In addition, there should be procedures or means in place to ensure that flights are correctly identified (tail number, date, and leg) in the FDM data. This is not only important for the FDM programme, but also for linking FDM data with data from other sources.

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<sup>13</sup> Refer to EOFDM document titled ‘Key Performance Indicators for a Flight Data monitoring Programme’. This document is published on EOFDM webpage.

<sup>14</sup> In the EU rules for air operations, AMC1 ORO.FC.A.245 specifies a data collection rate of at least 60 % for a new ATQP and 80 % for an extension to an ATQP.

*Note:*

Sufficient amount of flight data is needed to test an FDM algorithm definition (see also Chapter II). Adequate flight collection rate is important for this purpose. However, collecting enough flight data for testing purposes can be problematic for a small fleet or a rare event.

### Is there enough information in the recorded flight parameters?

The minimum list of flight parameters required for crash-protected flight data recorders (FDRs) on modern aeroplanes and helicopters can be used as a baseline for any FDM programme. This list can be found in the EU rules for air operations, in the AMC to Part-CAT, CAT.IDE.A.190 (FDR, aeroplanes) and CAT.IDE.H.190 (FDR, helicopters).

In case some flight parameters are not recorded by the quick access recorder (or equivalent):

- It might be possible to retrieve those flight parameters from the FDR. Using FDR recordings for the FDM programme is permitted, and some equipment manufacturers offer hand-held devices allowing the readout of the FDR on the aircraft. Keep in mind that the FDR is not intended for routine readouts and on some aircraft types it may be difficult to access the FDR. Furthermore, the FDR is a minimum equipment list (MEL) item affecting operational availability and it has a limited recording duration (usually not more than 25 hours). The use of the FDR for FDM purposes often requires dedicated procedures.
- Reconstructing missing flight parameters based on the recorded flight parameters is feasible, but may prove to be labour-intensive and it may produce inaccurate results. Possible errors introduced by the reconstruction method (noise, drift, offset) and the domain where the reconstructed flight parameter is valid should be carefully assessed, unless the reconstruction method has been approved by the aircraft manufacturer or is recommended by internationally recognised standard. In addition, it is important that the FDM tool clearly differentiates recorded parameters from reconstructed parameters.

The assessment of the available information in the recording should not be limited to the presence of a corresponding flight parameter. The information content of each flight parameter should also be reviewed and validated.

Example: If a discrete flight parameter is used to record TAWS warnings, it may mean that only a TAWS warning was triggered but it may not provide details of the TAWS modes and the severity of the TAWS alert (caution or warning).

### Is the flight data frame layout documentation clear and complete?

If a flight parameter is intended to be used for programming FDM events, it is essential to be certain about its meaning, source, units, and sign convention. All this information should be provided in the flight data frame layout documentation.

The source of a given flight parameter recorded for flight data monitoring might not be the same as the source of the information presented to the flight crew members.

Example: Only the flight parameters presented on one side of the cockpit might be recorded, or the recorded flight parameter could come from a source that is not presented as such to the flight crew members (for instance, the pitch attitude as provided by an inertial reference

system is recorded, while the pitch attitude displayed to the flight crew takes into account the values provided by all three inertial reference systems).

Sometimes the flight data frame layout documentation lacks clarity about the source of a flight parameter. For instance, a parameter is named ‘flap position’ and it is not clear if this means the flap/slats handle position or the actual position of the flaps. The FDM specialist should refer to manufacturer documentation or request the support of maintenance staff in that case.

*Note:*

ARINC specification 647A, Flight Recorder Electronic Documentation, provides an internationally recognised standard for the content and format of the flight data frame layout documentation.

Beside information on the flight parameter source, complete and correct information on the engineering unit, sign convention and/or format of a flight parameter is essential for using it in an FDM algorithm.

Example: the angle of attack parameter may be recorded in degrees, or as a ratio between the angle of attack and a reference angle of attack (for instance for stall angle of attack);

Example: the air/ground status parameter is usually a binary parameter. Depending on flight data frame layout, the value ‘0’ for this parameter may mean ‘on the ground’ or ‘airborne’.

### What does the quality of flight data allow for?

*Note:*

Guidance related to flight data quality is also provided in Chapter 8 of CAA UK CAP739.

Understanding the quality of flight data used in the FDM programme is essential to avoid designing FDM algorithms that work on paper and then fail during testing because of insufficient flight data quality.

#### *Flight parameter quality*

When considering a flight parameter used by the FDM software it should be correctly recorded and decoded (e.g. the sign convention is correct, the unit convention is correct). Typical quality problems affecting flight parameters are:

- Spikes – sudden variation of flight parameters to abnormal values;
- Malfunction – a flight parameter is not recorded properly throughout a flight due to a system malfunction (either known or unknown);
- Freeze – at a given point of a flight, a parameter ‘freezes’ and it is no longer properly recorded (usually due to a system malfunction);
- Offset/bias – there is a constant difference between the flight parameter value and the actual physical value. This may occur to flight parameters that require calibration.

*Note:*

The case of ‘outliers’ is discussed in Chapter III.

### *The flight-phase splitting logic*

Another source of flight data quality problems may be the flight-splitting logic or the flight-phase-splitting logic programmed in the FDM software. A correct detection of flights and a correct transition between two consecutive flight phases by the FDM software can be critical for the successful implementation of FDM algorithms. Any inconsistency should be investigated.

Example 1: An FDM event algorithm designed to capture rejected take-offs may not work if the software only recognises a flight when the aircraft is airborne.

Example 2: Detection of the transition from the landing phase to the taxi-in phase. In order to correctly detect taxi-in when the aircraft uses a high-speed taxiway to exit the runway with an aircraft speed above the usual taxi speed, a flight phase splitting logic should not only monitor the groundspeed to initiate the taxi-in phase. If not properly implemented, the detection of transition between landing roll and taxi-in might cause an FDM measurement algorithm monitoring the taxi speed to provide wrong results.

Example 3: Some FDM software might wrongly detect a flight during maintenance activities and generate undesired FDM events. False flight recordings should be flagged so that they do not pollute any statistical analysis.

### *Assessing the overall quality of the flight data*

Furthermore, the quality of the whole dataset needed for FDM-based analysis should be assessed and not only the quality of individual flight parameters or the reliability of the flight-phase-splitting logic.

Although there is no specific methodology for assessing flight data quality, general data quality standards can be applied to flight data. Six data quality dimensions could be considered for assessing flight data quality: accuracy, completeness, uniqueness, timeliness, validity, and consistency. Table 1 provides explanations and examples for these six data quality dimensions.

Table 1 — proposed data quality dimensions and examples of application to an FDM programme

<b>Data quality dimension</b>	<b>Proposed definition</b>	<b>Examples of data quality indicator applicable to an FDM programme</b>
<b>Completeness</b>	The proportion of stored data against the potential of '100 % complete' <i>Presence of non-blank values (%)</i>	<ul style="list-style-type: none"> <li>At least X % of flights performed with aircraft in the scope of the FDM programme are imported in the FDM software.</li> <li>Cuts (missing value %) in flights that are imported in the FDM software represent less than X % of the total duration of those flights.</li> </ul> <p><i>Note: it is important to analyse missing value patterns in the data to identify any potential system error.</i></p>
<b>Uniqueness</b>	No element will be recorded more than once based upon how that element is identified.	<ul style="list-style-type: none"> <li>Not more than X % of flights imported in the FDM software are duplicates (i.e. the same flight</li> </ul>

Data quality dimension	Proposed definition	Examples of data quality indicator applicable to an FDM programme
	<p><i>Inverse of level of duplication (%)</i></p>	<p>appears twice or more in the flight data).</p> <ul style="list-style-type: none"> <li>• Every flight parameter used by the FDM software is unique (no case where a flight parameter is decoded twice with different names, so that two flight parameters used by the FDM software contains the same information)</li> </ul>
<p><b>Timeliness</b></p>	<p>The degree to which data represent reality from the required point in time.</p> <p><i>Time difference between the real-world event being recorded occurred.</i></p>	<ul style="list-style-type: none"> <li>• Not more than X % of flights are 10 days old or older when they are scanned by the FDM software.</li> <li>• All flight parameters are recorded with an appropriate recording rate for the FDM events where they are used (e.g. the ‘gear compressed’ parameters are recorded at least once per second in order for an FDM event related to the take-off distance to be accurate enough).</li> </ul>
<p><b>Validity</b></p>	<p>Data are valid if they conform to the syntax (format, type, range) of their definitions.</p> <p><i>Data items deemed to be valid (%)</i></p>	<ul style="list-style-type: none"> <li>• Less than X % of the recordings of a continuous flight parameter used by the FDM software have values that are outside the range of possible values (e.g. when considering the groundspeed parameter, physically impossible values would be values above the design diving speed).</li> <li>• Less than X % of the recordings of a discrete flight parameter used by the FDM software have values that are not valid according to the format of that discrete flight parameter (e.g. the ‘flaps/slats handle position’ parameter should only take the values {0,1,2,3,4}, but several recordings of that parameter have the value 5, which is invalid because there is no corresponding position of the flaps/slats handle).</li> </ul>
<p><b>Accuracy</b></p>	<p>The degree to which data correctly describes the ‘real world’ object or event being described.</p>	<ul style="list-style-type: none"> <li>• The accuracy of each flight parameter used by the FDM software meets the accuracy target (e.g. when the aircraft is at the parking stand, the normal</li> </ul>

Data quality dimension	Proposed definition	Examples of data quality indicator applicable to an FDM programme
	<i>Percentage of data entries that pass the data accuracy criteria (%)</i>	acceleration parameters should not deviate from the neutral value – corresponding to gravity force – by more than X %. For flight parameters that are also required to be recorded on board the FDR, refer to EUROCAE ED-112A. <ul style="list-style-type: none"> <li>• The UTC time recorded in the flight data is accurate within X seconds.</li> <li>• The flight phase transition determined by the flight splitting logic of the FDM software is accurate within X seconds.</li> </ul>
<b>Consistency</b>	<i>Absence of difference when comparing two or more representations</i>	<ul style="list-style-type: none"> <li>• The values and the variations of flight parameters used by the FDM software are consistent with each other (e.g. when the recorded groundspeed increases, the recorded longitudinal acceleration indicates at a forward acceleration; when the aircraft banks to the right, the heading indicates at a right turn).</li> </ul>

Is the performance of recorded flight parameters enough to program effective FDM algorithms?

In this document, performance of a flight parameter means a set of characteristics such as:

- sampling rate,
- accuracy,
- recording resolution, and
- operational range.

The FDM specialist should have a good understanding of the systems or sensors generating flight parameters and of their intrinsic limitations (e.g. the calibrated airspeed is usually not valid under 40 kt, the radio-altitude usually does not operate above 2 500 ft height, the glide path of an instrument landing system is not linear close to the ground, etc.).

EUROCAE document ED-112A specifies minimum performance for the flight parameters recorded on the FDR. These specifications were copied into the EU rules for air operations, in AMC to Part-CAT, CAT.IDE.A.190 and CAT.IDE.H.190. This could serve as a baseline for flight parameters collected in the FDM programme.

Example: The minimum sampling rate specified in ED-112A for the landing gear compression parameters is 1 sample per sec.

In a dynamic flight phase where the values of a flight parameter are expected to vary rapidly (e.g. take off, go-around, avoidance manoeuvres), the accuracy of that flight parameter might be lower than in a stable flight phase. ED-112A specifies that data ‘shall be obtained from sources within the aircraft, which provide the most accurate and reliable information under both static and normal dynamic conditions.’. Hence, it is preferable to use the same flight parameters as those recorded on the FDR for FDM algorithms when they are implemented on dynamic flight phases.

On some aircraft models, the flight data frame layout may be reconfigured without a retrofit. This allows an operator to standardise the flight data frame layouts across the fleet, to modify sampling rate of some flight parameters and/or record additional flight parameters. Enhancing the flight data frame layout may generate savings for maintenance (by permitting a better monitoring of the condition of systems and engines) and/or for operations (e.g. fuel savings).

Example: A flight parameter recording the status of passenger doors may be correlated with the scheduled arrival time for accurately computing the delay at arrival of a given flight.

*Note:*

It is important to understand the limitations of discrete-time signals (which recorded flight parameters are) and the effects of signal sampling. It is not within the scope of this document to explore these aspects in detail, but generically speaking, if the sampling rate of a signal is close to or lower than the signal’s highest possible frequency, then sampling may result in significant loss of information. In the case of a flight parameter, too low sampling rate may lead to erroneous results and/or missed FDM events.

*Note:*

Updating the flight data frame layout may have an unexpected impact on already-implemented FDM algorithms (see Section I.3).

### 3. FDM algorithms

#### Identifying an initial set of FDM algorithms

Initially, it is recommended to implement a small set of ‘broad’ FDM algorithms (FDM event algorithms and/or FDM measurement algorithms) covering all flight phases and the main categories of aviation occurrences, and then to progressively complete this set with ‘specialised’ FDM algorithms that address more peculiar operational safety risks.

- ‘Broad’ FDM algorithms are usually those that are designed to help capture unsafe outcomes, such as a risk of CFIT or of runway excursion. A broad FDM algorithm may potentially capture all occurrences related to a given operational safety risk.

Example: The FDM event algorithm<sup>15</sup> ‘risk of runway overrun at landing’ detects that during the landing roll the aircraft is approaching the end of runway too fast (i.e. the groundspeed seems too high for the remaining distance). This FDM event algorithm is likely to reliably capture all landings with a higher risk of runway overrun.

Broad FDM algorithms are helpful for the initial assessment of safety issues. However, because a broad FDM algorithm is rather unspecific, the output data generated by such an algorithm need to be analysed to sort it out into smaller categories. A broad FDM algorithm is therefore not very suitable for monitoring specific safety issues, but can be a good starting point.

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<sup>15</sup> Examples in this section refer to FDM event algorithms, but the concepts of broad FDM algorithms and specialised FDM algorithms are also applicable to FDM measurement algorithms.

Example: It would be desirable to sort out individual detections by the FDM event algorithm 'risk of runway overrun at landing' into subcategories such as 'touchdown point far beyond the recommended touchdown zone' and 'too little/too late deceleration after touchdown'.

- 'Specialised' FDM algorithms are helpful for detecting specific deviations from the SOPs or accepted practice, even if they do not necessarily result in any unsafe outcome.

Example: A specialised FDM event algorithm could detect excessive groundspeed at landing (groundspeed is too high when the aircraft is passing the landing threshold). If the groundspeed at landing is excessive on a runway much longer than the landing distance, there is no immediate risk associated with this deviation.

A specialised FDM algorithm may be useful to track a specific safety issue e.g. related to a piloting technique or to the implementation of a particular SOP. However, the scope of a specialised FDM algorithm is rather narrow, so that unforeseen safety issues may not be detected if only specialised FDM algorithms are implemented in the FDM programme.

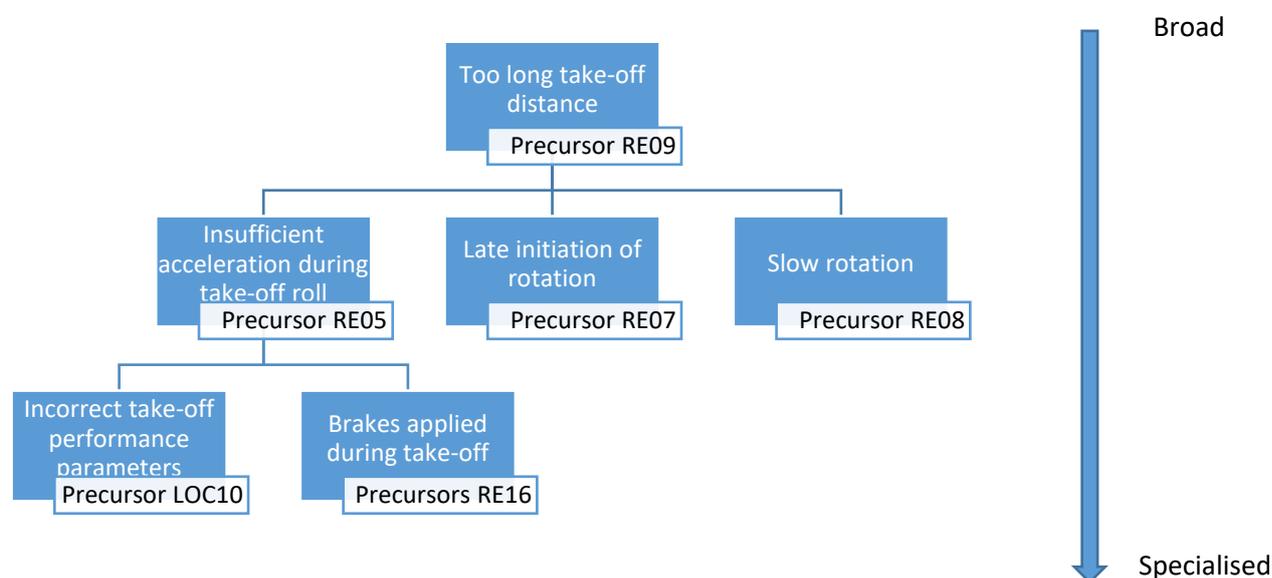
Example (see also Figure 1):

An excessively long take-off distance may have various causes, such as:

- insufficient acceleration during the take-off roll, which may be caused by
  - inadequate take-off thrust setting, or
  - the inadvertent application of brakes during the take-off roll; or
- the late initiation of the rotation; or
- a too slow rotation.

In this example, an FDM algorithm monitoring the take-off distance could be considered as 'broad' while an FDM algorithm capturing information about the application of brakes during the take-off roll is rather 'specialised'.

Figure 1 — example of organising FDM algorithms according to possible causal relationships and whether they should be considered broad or specialised



### Using pre-defined FDM algorithms

Most FDM software suites contain pre-defined FDM event algorithms and/or pre-defined FDM measurement algorithms, and it is tempting to rely on such pre-defined algorithms. In such cases, it is critical to understand how these algorithms were designed and to subject them to thorough testing (see Chapter II). This is necessary to check that these algorithms really capture what they are expected to capture (limit the proportion of ‘false negatives’) and that they do not generate too frequent undesired FDM events (also called ‘nuisance events’ or ‘false positives’).

Note: the operator is responsible for the quality assurance of the FDM solution they use, as there is no FDM software certification process.

A pre-defined FDM algorithm should be sufficiently documented by the FDM software provider to address such questions. In addition, the trigger logic of a pre-defined FDM event could be consulted with flight operations and flight training departments, to confirm that it is consistent with the operator’s SOP and accepted practice. The significant experience of instructor pilots (flight, simulator, theoretical, etc.) is a valuable resource for this purpose.

If the operator does not have access to all programming details, they should, as a minimum, be provided with a detailed description of the algorithm logic for each pre-defined algorithm. Otherwise, it is recommended to (re-)define the FDM algorithm from scratch. This may require assistance from the FDM software provider.

Example 1: What ‘time’ do the following FDM measurement algorithms actually measure?

- ‘rotation time’ measurement algorithm does it measure the time to reach the lift-off pitch or the time until the aeroplane actually lifts-off, or something else?

- ‘take-off distance’ measurement algorithm: does the measurement start when the aircraft turns onto the runway heading, or when a given power setting is selected, or other criteria?
- ‘TAWS alert’ event algorithm: what are the TAWS modes captured? And is the trigger logic of this algorithm consistent with the specific implementation of TAWS on the aircraft (for instance: is the bank angle threshold of the algorithm equal to the actual bank angle value that will trigger a ‘bank angle’ TAWS warning on that aircraft?).

Example 2: FDM event algorithm detecting the condition ‘unstable approach’: what are exactly the deviations captured by this algorithm?

It is often necessary to adjust pre-defined FDM algorithms to specific operation. Unfortunately, some FDM software offers only limited functionality to modify pre-defined algorithms. In addition, some operators may not have sufficient internal knowledge to re-design these algorithms.

An FDM algorithm may work for a given aircraft model but not for another one, even when they belong to the same family (for instance, because the applicable trigger value is not the same). In addition to this, when aircraft have different flight data frame layouts (even if they are from the same make and model), this might result in an FDM algorithm producing inconsistent data across that fleet. Therefore, FDM algorithm testing should be performed for each flight data frame layout.

#### Maintaining the set of FDM algorithms

The safety risk register of an operator changes over time, with new safety threats emerging and the frequency and severity of any given safety risk increasing or decreasing. This means that the FDM event algorithms and FDM measurement algorithms should be checked against the operational safety risks of the safety risk register at regular time intervals.

Examples of changes that could generate new operational safety risks:

- Operating on a new airfield or runway,
- Major changes in cockpit training,
- Change of operation area,
- Including new types of aircraft in the fleet, or
- New or revised SOPs.

## II. Defining, testing, and validating an FDM algorithm

The setting of an FDM event algorithm or an FDM measurement algorithm involves three different phases, which can be identified as:

- Definition
- Testing
- Production
- Updating

### 1. Defining an FDM algorithm

#### Identifying the necessary data

In the phase of the definition of the FDM algorithm (FDM event algorithm or FDM measurement algorithm), some initial assessment should be performed.

Based on the description of the operational safety risk (see Chapter I), define which event should be captured through FDM.

Verify whether the data frames contain all the necessary flight parameters to address the event to be captured. Verify if the frequency and the recording resolution of each needed flight parameter are sufficient<sup>16</sup>.

Example: Case of an FDM event algorithm for detecting structural limit exceedance case such as caused by overweight landing. During the initial assessment period the flight parameters that will be used and the associated aircraft limitations should be identified in cooperation with the engineering department.

Identify those additional flight parameters that could be helpful for testing and validation. The FDM algorithm should also include the definition of a data snapshot.

#### *Note:*

If other data sources are available, which could be automatically linked with the FDM data (air safety reports (ASR), weather, navigation data, etc.), it is advisable to already assess at this stage what data could be used to enrich the data snapshot of the FDM event<sup>17</sup>.

Consider whether the FDM software uses automatic flight splitting or flight phase splitting. This may have significant influence on the effectiveness of the FDM algorithm. See also Section I.2.

#### Verifying the flight parameters

For those flight parameters that have not yet been verified, check that their values, sign, and variations are consistent. Verify whether there are some problems affecting flight parameter quality (refer to

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<sup>16</sup> Some parameters may be calculated from those recorded in the data frame, but care should be taken in this process. A typical example is computing the variations of a recorded parameter in order to assess its derivative, i.e. calculating the pitch angle rate from the recorded pitch attitude angle. This computation introduces noise. To cope with this, filtering could be added after performing the derivative calculation.

<sup>17</sup> Refer also to EOFDM document 'Breaking the silos', Chapter I.

Section I.2) that may have a negative effect on the final result. Start by examining flight parameters plots for some sample flights. This is not an exhaustive check and some problems with the flight parameters may only become obvious at a later stage.

Most FDM software allows to quickly review parameter values distributions over large amounts of flights. This should be used as much as permitted by the available data and the computational power of the software. As a minimum, such review should include looking for out-of-range values, for outliers and extreme values, for missing values, abnormal distributions, etc.. Scatter plots and boxplots are also helpful (see Chapter III).

Check whether the performance of some flight parameters may affect the accuracy of the trigger logic or limit its domain of validity.

Examples:

- The airspeed is usually based on total pressure measurements provided by a pitot probe and therefore the airspeed is not accurate when the aircraft is taxiing at lower speed;
- If the latitude and longitude parameters are provided by an individual inertial reference system, they will drift over time and provide an inaccurate position toward the end of the flight;
- The difference between the magnetic heading and the true heading is usually of several degrees at high latitudes, so that magnetic declination correction may be needed before using the magnetic heading.

*Note:*

It is also advisable to verify the quality of the data sources other than flight data.

Example: for computation of height at threshold, the position of the runway threshold is necessary, and therefore an up-to-date database of airports and runways is required for an accurate computation.

The following illustrates flight parameters checks that are recommended during the development steps of an FDM event algorithm with the example of an algorithm monitoring flap speed limit exceedance:

1. The first step ensures that the flight parameters needed are recorded: airspeed and aircraft flap configuration parameters.
2. Next, the recording rate and recording resolution of each of these flight parameters are checked, to ascertain if they can deliver the desired output. For instance, if one of these flight parameters is recorded only once every 4 seconds, this can have a considerable effect on the results. When an FDM algorithm relies on flight parameters with different recording rates, down-sampling and/or interpolating the values will be necessary, but, if not done properly, this might result in a misleading outcome.
3. Then, a quality check is performed. That is, if other flight parameters are recorded, such as airspeed from other aircraft sources, they are used to increase the robustness of the FDM event algorithm.
4. Then, a data snapshot of other relevant flight parameters values to be extracted when the FDM event is detected, is defined: altitude, rate of climb/descent, landing gear position, etc. These extracted values are checked for consistency.

## Defining a search window

Defining a ‘search window’ (i.e. conditions where the FDM algorithm is applicable) usually accelerates data processing by reducing the amount of flight data to be processed by the FDM software.

Example: An FDM measurement algorithm monitoring the taxi speed does only need to be applied to those parts of the recording where the aircraft is on the ground. The search window specifications of such an FDM event could be:

- ‘only applies to a taxi phase’ (if flight phase splitting is provided by the FDM software), or
- ‘only apply when the value of the air-ground status flight parameter is “ground”’.

## Defining the trigger logic of an FDM event algorithm

To define the trigger logic of the FDM event algorithm, the ‘normal’ domain needs to be established. Many limitations can be obtained from the AFM. In addition, the operator’s SOPs define operating limits. When no indication is provided by available documentation and SOPs, a statistical distribution of relevant FDM parameters may also be helpful to define the range of normal operation. Known reference flights within the flight data may also be useful to help define the ‘normal’ domain.

If feasible, an algorithm that frequently triggers is preferable for the first stage of testing since it can be tested with a limited amount of flight data. For instance, TAWS cautions are more frequent than TAWS warnings and therefore a smaller flight data sample will be needed for testing the trigger logic.

To make the trigger logic more robust, a ‘confirmation time’ may be introduced: the FDM event algorithm does not detect an event every time the trigger condition is met at a single point in time, but only if it is met over a period exceeding a given duration (often designated as ‘confirmation time’, ‘persistence time’, or ‘time filter’). Another possibility to make the trigger logic definition more robust is to base it on several simultaneous conditions (confirmation by several conditions).

Example 1: FDM event algorithm detecting excessive taxi speed (i.e. ground speed exceeding the maximum taxi speed permitted by the taxi SOPs). The introduction of a confirmation time of X seconds will make the trigger logic insensitive to spikes and very short deviations, while an excessive taxi speed sustained during X seconds will trigger an FDM event.

Example 2: FDM event algorithm detecting TAWS cautions – double validation using the parameters corresponding to TAWS aural and visual signals. The FDM event is triggered only if the TAWS aural caution is active and the TAWS caution light is on.

Example 3: FDM event algorithm detecting a high torque or a high rotation speed of the main rotor of a helicopter, such an FDM event usually requires a confirmation time.

## Defining severity levels for FDM events

As explained in HeliOffshore good practice document on helicopter FDM<sup>18</sup>:

‘It is important to draw a distinction between the severity of an individual event and the operational risk that event presented. Severity levels are assigned according to a numerical algorithm (in order to

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<sup>18</sup> HeliOffshore — Helicopter Flight Data Monitoring (HFDM) Recommended Practice for Oil and Gas Passenger Transport Operations, Version 1.0, September 2020 (HO-HFDM-RPv1.0)

assist analysts by highlighting certain events) but a high severity event does not imply high operational risk. [...]

Moving from event severity to event operational risk involves a further step in which an analyst assesses the full context around an event. This process may be supported by additional procedures including review groups, written criteria or a risk assessment matrix’.

Hence, the following distinction should be kept in mind:

- *‘FDM event severity level/score’* means a level/score that is allocated to an FDM event according to pre-defined criteria, and that is determined based on FDM data. The FDM event severity levels/scores may be used for supporting risk assessment, but FDM event severity levels/scores taken alone are not indications of actual risk.
- *‘assessed event risk level’*= the assessed level of risk (of an undesired outcome e.g. catastrophic, hazardous) for a given event. This event may have been detected through FDM or not (e.g. a laser attack or a near-collision with a vehicle on the ground will leave no clear trace in the FDM data). The risk level should be assessed in view of all available data (occurrence report, FDM, weather data, traffic data, etc.), it should preferably follow a risk assessment method, and, where necessary, involve expert knowledge. Because of this, risk level assessment cannot be fully automatised. Assessing the risk level of events is part of the operator SRM, not just an FDM process.

When several severity levels are applied to FDM events, they should be defined according to a severity level classification that is consistently applied across all FDM event algorithms. Otherwise, it will be difficult to make meaningful indicators from FDM events.<sup>19</sup>

There can be many appropriate severity level classifications and Table 2 just provides an example.

In addition, FDM event algorithms could be used to produce either ‘leading’ or ‘lagging’ indicators, depending on their definition or of their threshold values. For example, a leading indicator for the risk of mid-air collision could be based on monitoring the vertical rate during climb, while a corresponding lagging indicator would detect level busts.

Another example is the detection of hard landings. Assuming that the normal acceleration limit defined by the aeroplane manufacturer for a given aircraft model is 2.1 g, the FDM event algorithm could be designed in two ways:

1. By setting the threshold values below 2.1 g, an information about how close it is to the hard landing limit is produced. This can be qualified as a leading indicator.
2. By defining threshold values above 2.1 g, an information about by how much the hard landing limit has been exceeded is generated. The main advantage of this approach is that it only extract events that require an action according to the maintenance instructions. But when the FDM event algorithm triggers, the undesirable situation (hard landing, structural damage, etc.) has already happened. This can be qualified as a lagging indicator.

Both approaches have their advantages and drawbacks.

*Note:*

Grouping flights according to an event severity classification may limit subsequent analyses (e.g.

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<sup>19</sup> If the flight data are also used for continuing airworthiness purposes, a severity level classification following different principles may be needed, or even in some cases severity levels may not be relevant.

because it might exclude certain flights), and the FDM specialist should remain aware of this. FDM measurements may be more appropriate for finer analyses.

Table 2 — example of a severity level classification applicable to FDM event definitions

Level	Description	Notes	Examples
0	<p>Thresholds at Level 0 should be:</p> <ul style="list-style-type: none"> <li>indicative of a deviation from SOP and/or accepted practice; and</li> <li>specific for the operation (fleet and/or operator) based on the equipment and operating environments.</li> </ul>	<p>Level 0 does not necessarily constitute a safety hazard; however, monitoring trends at this level can be used to:</p> <ul style="list-style-type: none"> <li>measure SOP adherence;</li> <li>compare an operation to the aggregate to identify outliers or potential areas of concern;</li> <li>assess the potential need to re-evaluate and/or update SOPs.</li> </ul>	<p>Momentary and limited exceedance of the recommended vertical speed during the approach.</p>
1	<p>Thresholds at Level 1 are indicative of having approached or exceeded at least one primary risk mitigation.</p>	<p>Level 1 events may occur frequently enough to provide statistically significant trends within an FDM program (depending on the volume of operation). Trends at this level should have mitigation plans and not just be monitored.</p>	<p>TAWS ‘sink rate’ alert</p>
2	<p>Thresholds at Level 2 are:</p> <ul style="list-style-type: none"> <li>indicative that safety margins have been consumed or reduced to the final safety barrier;</li> <li>at levels significant enough that a crew would likely notice the event and generate a voluntary safety report.</li> </ul>	<p>Level 2 events should be rare. A root cause analysis would typically be performed on each individual Level 2 event.</p>	<p>TAWS ‘pull up’ alert</p>
3	<p>Thresholds at Level 3 are indicative that the undesired state occurred.</p>	<p>Level 3 events should be extremely rare. Thresholds at Level 3 may have values set at points beyond which protection systems are intended to have activated.</p> <p>A full operator (and possibly State) investigation would typically be performed on each individual Level 3 event.</p>	<p>TAWS ‘pull up’ alert and the radio-height parameter show that the minimum height was less than X feet, i.e. this was a near collision with the terrain.</p>

## 2. Testing an FDM algorithm

After the FDM algorithm has been defined, some coding and testing has to be done (test phase). Statistics and analysis about this algorithm should never be shared with third parties before it is confirmed to work correctly.

### General considerations

*Note:*

Some of the considerations provided in this section are focussed on FDM event algorithms. Other considerations address ‘FDM algorithms’, i.e. they are applicable to FDM event algorithms and to FDM measurement algorithms alike.

During the testing phase, the operator could use archived records in its occurrence reporting system to identify flights that are relevant for a new FDM event algorithm to be tested. Such flights can for instance help to validate the trigger logic of that algorithm.

*Note:*

The occurrence reporting system may include more than just flight crew reports. Maintenance reports and operational flight plans could be also collected.

For testing purposes, the threshold value could be changed to trigger an FDM event for parameter values that are frequently encountered in operation. After verifying that the FDM events are valid, the threshold value could then be set to the appropriate value<sup>20</sup>.

Check flights and maintenance flights are likely to generate FDM events therefore they could be used for testing the trigger logic of a new FDM event algorithm.

It could also be useful to build up a ‘control set’ composed of flights that do not contain the events that a tested FDM event algorithm is meant to detect, and that possibly contains flight sequences or flight manoeuvres that could trigger a false positive. This control set would be useful for ensuring that the tested algorithm does not generate too frequent false positives.

With regards to the data used for test, there are two options:

- When it is technically possible, it is recommended to test on a separate platform which contains a sufficient number of historic flights, to ensure that operational statistics are not compromised. This usually allows to complete the testing faster. However, not all FDM software allows this option.
- If testing on a separate platform is not possible, the testing will need to be performed on ‘live’ flight data. In that case, the testing process could take a little longer depending on the amount of data received every day. The potential impact on the live data processing systems should be carefully considered, as it could affect historical analyses. However, this option may give more reliable results because the test has been performed on recent flight recordings.

*Note:*

When considering two aircraft of the same make and model that have different engine models or slightly different airborne equipment, the same FDM algorithm may produce different results.

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<sup>20</sup> EOFDM WG-B document ‘Guidance for the implementation of FDM precursors’ provides guidance on setting a threshold.

Therefore, it is important to test the FDM algorithm on all flight data frame layouts implemented on the operator's fleet.

Before an FDM algorithm is put into production (see Section 3), its testing results should be reviewed and approved. The way this is organised depends on the internal organisation of each operator, but it is recommended to establish a formalised approval process. To ensure that the FDM algorithm definition is consistent with the operator's SOP and what is considered acceptable piloting skills, it is also helpful to include representatives of flight operations and flight training department in the approval process.

Also, according to EU rules for air operations, the FDM programme is part of the operator's SMS and the safety manager should be responsible for the FDM programme<sup>21</sup>. Because of this, the safety manager should verify whether new FDM algorithms are appropriate and relevant for addressing the operational safety risk. This verification should be done before the algorithms are put into productions.

#### Example of a testing plan for an FDM event algorithm

1. Test the FDM event algorithm against recordings of flights corresponding to known events (e.g. reports from the occurrence reporting system) to verify that the FDM event algorithm triggers for most of those flight recordings, that is to say the false negative rate or rate of 'missed FDM events' is low;
2. Test the FDM event algorithm on a sample of flights corresponding to a week of operation, or a month of operation, so that a rate of nuisance FDM events per time period can be estimated; this rate has a direct impact on the workload of manually validating FDM events.
3. Incrementally increase the sample size and the number of aircraft models in the scope of the FDM event algorithm to progressively gain confidence in the trigger logic. The sample size should be sufficient to establish reliable statistics of nuisance FDM events.

#### *Note:*

If the new FDM event algorithm is linked to a specific departure or destination (e.g. an airfield with peculiar procedures due to terrain or obstacles), the sample should obviously include these.

#### Case of a low volume of operation

If the fleet size is small or the flights are performed at irregular time intervals, it may take longer to build a suitable data set to test the FDM algorithm. If aircraft utilisation is low, effective testing may not be possible, because the amount of data produced by the FDM algorithm is likely to generate unreliable statistics.

In that case, each flight may have to be reviewed in detail. To support analysis, a small set of flight parameters could be collected at specific points, to progressively gain confidence in the FDM algorithm.

In addition, certain types of maintenance or aircraft acceptance flights can provide a wealth of useful capture points, to test event sets against.

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<sup>21</sup> For aeroplanes, refer to Part-ORO, ORO.AOC.130 and AMC1 ORO.AOC.130.

With regard to analysing small data sets, refer to Chapter III.

### 3. Production phase

After successful testing and the FDM algorithm is considered compliant with the definition, it can be used for daily operations. This is defined as the production phase.

Despite testing, nuisance FDM events can still occur in production. This may result of sensor failure or other conditions that could not be anticipated during the testing phase. Therefore, a validation of the FDM events is necessary so that nuisance FDM events can be removed from the final statistics and the subsequent analysis process uses reliable data. This can be achieved by flagging each FDM event as valid or invalid. A well-defined data snapshot is helpful for a quick validation of the FDM events.

For high-severity FDM events that are not expected to be frequent, this validation should be done for every FDM event.

In the end, any FDM algorithm that is in production should be properly documented. Documentation should include:

- reason(s) for implementation including detailed description of the safety risks that are intended to be monitored, and a link to the safety risk register of the operator;
- date of implementation;
- dataset(s) to which the FDM algorithm was applied (e.g. was it only applied to newly received data or also to some of the historical datasets?);
- (for an FDM event algorithm) the threshold values and the rationale for selecting these threshold values; and
- any other comments deemed relevant for the event programming.

This will ensure that pertinent details will be kept over time, assisting the interpretation of the output of the FDM algorithm. This documentation can be provided in a dedicated manual, in controlled documents or even as comments in the programming code of the FDM algorithm.

### 4. Updating

The thresholds of some FDM event algorithms such as those based on SOPs may need to be modified from time to time. As such, a close interaction with the Flight Operations Department is paramount, so that any change to the SOPs can be incorporated in the FDM algorithms as soon as possible, otherwise the output of the FDM algorithms will lose their significance. Furthermore, a so-modified FDM algorithm should be run on the flight data received since the new SOPs became effective.

Finally, all the event algorithm documentation shall be amended containing:

- the reason(s) for any change, including detailed description of the new safety risks that are intended to be monitored, and a link to the safety risk register of the operator;
- the change(s) made to the programming (e.g., modified trigger logic or thresholds, old-new variable encoding, metric changes etc...); and
- the date of this implementation.

This will permit to track the history of changes performed to an FDM event algorithm, which will in turn assist interpretation of the related FDM events.

## 5. Examples

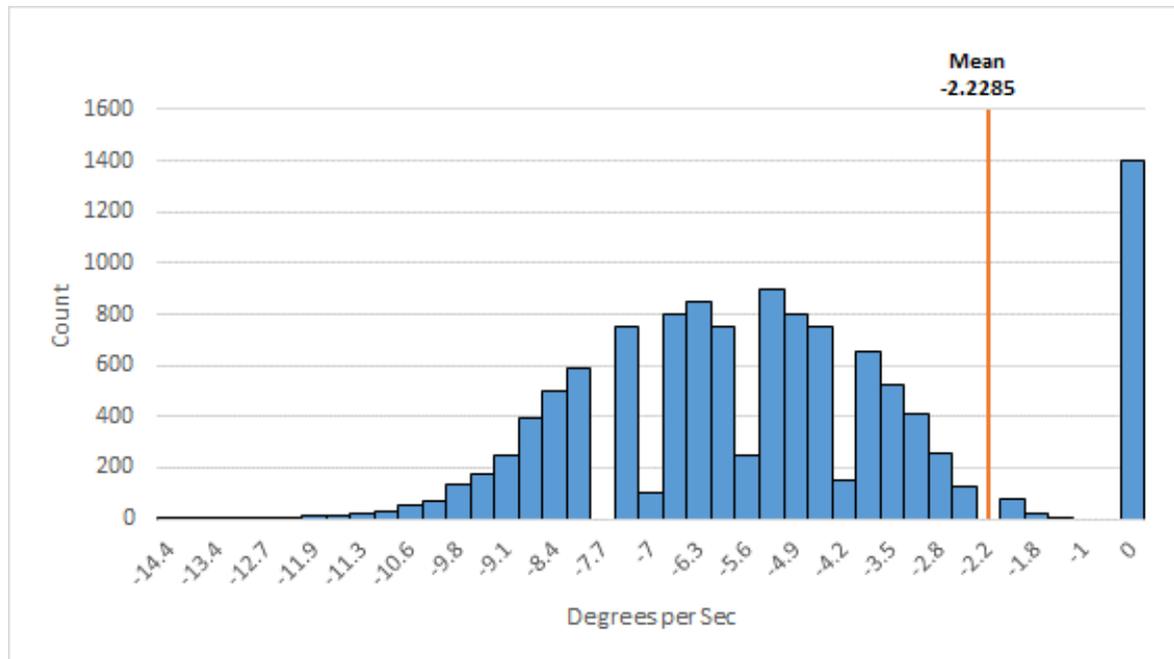
### Example 1: Incorrect capture window

The trigger logic below, originally used the groundspeed parameter to trigger an FDM event to measure the pitch rotation rate at take-off for helicopters.

*Where GROUND SPEED <= 15 Knots, measure greatest change in pitch attitude*

When this trigger logic was used on an aircraft that did not have a groundspeed parameter, many flights had a pitch rate of 0 degrees per second: see Figure 2. This resulted in a large number of invalid FDM events as well as incorrect statistical information.

Figure 2 — distribution of pitch rate at take-off that shows incorrect values

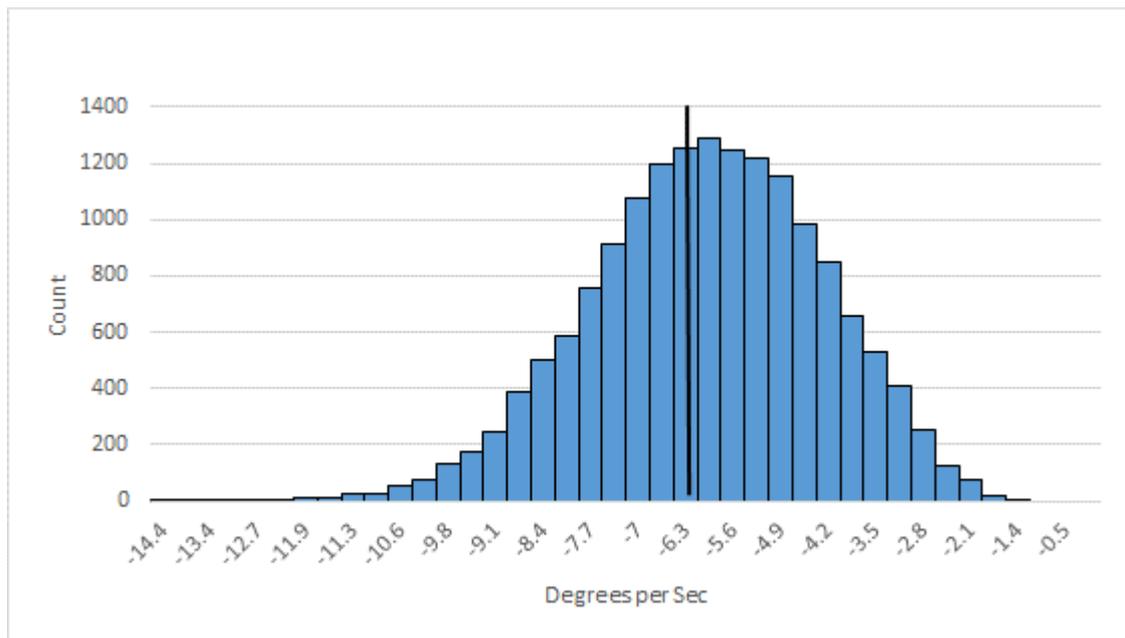


The trigger logic was corrected to use indicated airspeed parameter and the capture window was expanded to the following:

*Where INDICATED AIRSPEED <= 40 Knots, measure greatest change in pitch attitude*

The revised trigger logic now captures the correct pitch rate and reduces the number of invalid FDM events (see Figure 3).

Figure 3 — distribution of pitch rate at take-off

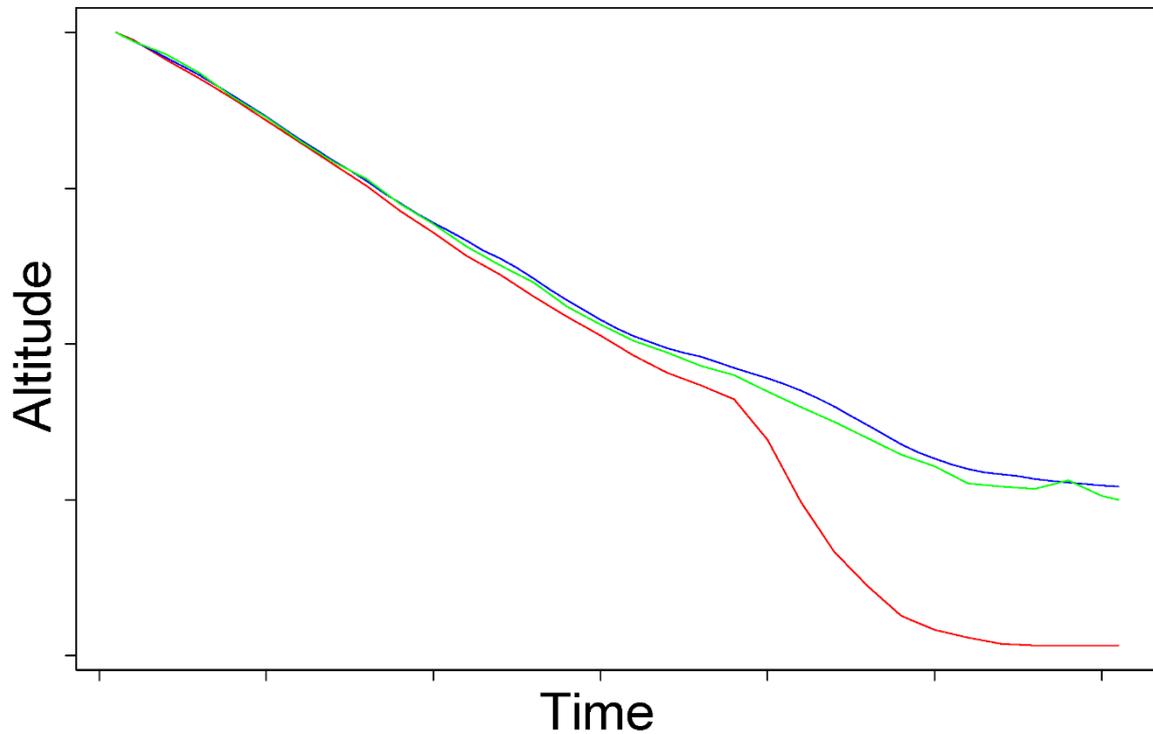


Example 2 – Use of the radio-altitude in helicopter FDM programmes

Figure 4 shows the evolution over time of some flight parameters related to a helicopter approaching an offshore installation. Pressure altitude and the integral of vertical speed are shown in blue and green and the radio altimeter (radalt) parameter is shown in red.

The three values show good agreement until the clear step in radalt value that occurs as the helicopter moves over the helideck. Any FDM algorithm using a radalt parameter in this situation must take account of this effect. For instance, differencing the radalt parameter to synthesise a vertical speed would produce erroneous values when the helicopter lands on a helideck.

Figure 4 — pressure altitude (blue), integral of vertical speed (green) and radio-altitude (red) of a helicopter approaching a helideck



### Example 3 – Monitoring taxi speed during turns

Many manufacturers recommend limiting the taxi speeds during turns. E.g. the A320 FCTM recommends a speed of less than 10 kt during turns of over 90°. Boeing has a similar recommendation for the B737NG. When defining an FDM measurement algorithm to capture speed during turns, the algorithm should compute heading change within a specific time interval. Otherwise, the algorithm might identify a turn of over 90° at almost every flight cycle.

### III. Producing meaningful FDM statistics and interpreting them

*Note:*

The ambition of this chapter is limited to introducing a few statistical tools and practical examples of their application to FDM data. Readers should keep in mind that explanations and examples in this chapter have been simplified and are incomplete. Many publications are available on the internet for further learning on statistics.

#### 1. Why use statistics?

Much of the benefit of FDM comes from analysing very large numbers of flights to look for trends in the operation. Statistics can help to achieve this. Statistics can also help to overcome biases.

When working with large amounts of data, statistical analysis methods are beneficial to highlight useful information. By its nature, any FDM program involves working with large amount of data, thus using correct statistical analysis methods is important to look for trends and emerging operational risk areas.

In general, statistics can be separated into two branches – descriptive and inferential. Inferential statistics draw conclusions about a whole population by studying a sample of that population, whereas descriptive statistics looks to summarise, collect, analyse and present a complete set of data. Inferential statistics are helpful especially if the flight collection rate is low. Descriptive statistics might be more appropriate if a high flight collection rate is achieved for all individual aircraft and routes.

*Note:*

More advanced predictive analysis (e.g based on machine learning) is outside the scope of this document.

The type of overarching questions that may be addressed using statistical analysis include:

- What has changed in the operation over certain time intervals?
- Is there a real trend, pattern, or correlation when considering a given FDM algorithm, airfield, runway, fleet, or pilot? How likely is it that a variation in the number or rate of FDM events just results from chance and not from a trend?
- Are there indications of the possible reasons for an observed trend or pattern?
- How can the information obtained from statistical analysis be turned into actionable solutions?

#### 2. Considerations before engaging into statistics

Before applying statistics to the FDM programme, the context of the flight data should be understood, as this may affect the relevance of any statistical computation. Below are a few questions that are worth considering:

- Which individual aircraft are included in the FDM programme, and since when? Long-term statistics cannot be produced for recently introduced aircraft.
- Are flight data collected from non-revenue flights (e.g. ferry flights, training flights, and maintenance check flights), or solely from commercial flights? Beware that invalidating FDM

events does not remove the flight data from non-revenue flights and these data could pollute future statistics if the flights are not properly identified as non-revenue flights.

- Are some areas of operation or airfields under-represented in the data collected by the FDM programme? Is this because of a technical reason (e.g. recovering flight data from the aircraft is challenging for some areas of operation)?
- What is the average time between the flight taking place and the data being analysed? What is the maximum time?
- Is the number of flights sufficient for statistical relevance and maintaining confidentiality? A minimum should be defined.

In addition, any FDM algorithm that is used for statistics should have been properly tested (refer to Chapter II). In the case of an FDM event algorithm, most nuisance FDM events of that algorithm should have been removed from the FDM events set prior to statistical analyses.

Furthermore, it is essential to clearly state the objective of a statistical calculation from the beginning. In particular the questions to be addressed by that calculation should be written down, as well as any simplification or assumption that was made to address those questions.

Example: In the last three months, there seem to have been more cases where the flight crews requested priority landing for being short of fuel.

Question: how often has the remaining fuel quantity at the time of touchdown been less than the final reserve fuel<sup>22</sup> as prescribed by the SOPs?

Assumptions: on the considered aircraft model, the fuel quantity is computed by a fuel quantity indication system, based on values taken from probes in each tank. The tolerance of the fuel quantity indication system as indicated by the aircraft manufacturer is '± (1 % of current FOB + 200 kg)'. This means that the expected accuracy of the fuel quantity parameter will probably not be better than 200 kg, and that this parameter will be less accurate at the beginning of the flight, when there is more fuel on board, than at the end of the flight.

One should keep in mind the simplifications and assumptions made for a statistical analysis when assessing the results. If some simplifications or assumptions are ignored during this assessment, this can lead to inaccurate or even wrong conclusions, and in turn, to inadequate decisions

It is also important to ensure that any comparison is 'like for like'. For example, a change in season may bring a change in flight hours (which is accounted for by using the FDM event rate), but it may also change the routes being operated (which may introduce other differences).

At the time this document was issued, most commercial FDM software offered only limited statistical analysis capability. In addition, as explained in EOFDM document 'Breaking the silos', a safety analyst needs to be able to bring FDM data together with other data sources (within a clear and robust framework for protecting sensitive and personal data). In this context, it is more important that the FDM software is capable of exporting output to formats that are compatible with other tools (such as numerical-analysis software or business intelligence tools) rather than having complete statistical data suites embedded in the FDM software.

*Note:*

Particular precautions regarding the sharing and dissemination of FDM-based indicators may be

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<sup>22</sup> According to ICAO Annex 6 Part I, the final reserve fuel for a turbine-engined aeroplane is 'the amount of fuel required to fly for 30 minutes at holding speed at 450 m (1 500 ft) above aerodrome elevation in standard conditions.'

needed in the case where the limited number of flights makes it possible to correlate such indicator with identification data from external sources (e.g. scheduling system) and to re-identify flight crew members.

### 3. Basic measures based on FDM events

#### Total number of FDM events or 'FDM event count'

The FDM event count is a basic metric that can be used as a starting point for evaluation. However, the absolute number of events is of limited use since a change in total number could be due to a change in event rate or a change in flying hours or activity. Therefore, a normalisation process is frequently considered.

#### FDM event rate

To compensate for changes in the volume of activity, calculating the FDM event rate is often used. To calculate the event rate of an FDM event algorithm, sum up the number of detections of that event (e.g. low rotation rate at take-off), and divide by the number of flights analysed. Rates are often quoted as a value per 1 000 flights or per 10 000 flights (or flight hours) to make the numbers manageable.

#### *Note:*

The format of the FDM event rate should be appropriate for tracking trends. Too small numbers might not catch the attention they deserve, therefore presenting the FDM event rate per 1 000 flights or per 10 000 flights is more appropriate than per 100 flights.

If only a limited number of flights is available for analysis, the FDM event rate will be extremely sensitive to small variations in numbers of FDM events (e.g. the influence of a single event on a data sample of 1 000 flights compared to 100 000 flights is a hundred times larger).

Every individual FDM event that indicates a severe deviation should be verified, and if it is genuine, analysed. Prioritisation between FDM events should not only rely on their rate, but also on their severity.

Although the use of FDM event rates can compensate for changes in the volume of activity, variations of the FDM event rate may be caused by many other factors such as meteorological conditions, route differences, aircraft type differences, etc. In this case it may be advisable to group the data by key factors such as fleet type, route, landing or take-off runway, etc.

Rates should be based on number of analysed flights instead of number of flights flown (often readily available from an operational schedule). This is important to eliminate statistical errors introduced by collection rate fluctuations (e.g. due to technical failure of flight data equipment or because of missing recording equipment on a specific fleet).

#### Trends over time

It may be useful to track an FDM event rate over time to look for any developing trend: see Section III.6.

### Including the severity level

Since an FDM event may correspond to a more severe or less severe event, the severity level of each FDM event should be taken into account when monitoring trends over time. The notion of severity level is addressed in Chapter II.

FDM events can be grouped by severity level. This allows monitoring a trend over time for events of the same severity level.

Example: A less frequently operated airfield has an overall FDM event rate (per approach) similar to those of other airfields, but it has comparatively a higher rate of FDM events with higher severity.

Another approach is to associate a weighting factor to each severity level. This allows aggregating FDM events of various severity levels.

*Note:*

The values of weighting factors may have a significant influence on aggregated severity scores. Therefore, these values should not be selected randomly, but based on a rationale.

Tables 3 and 4 illustrate how this could be used to allocate a single score for FDM events triggered during a given month of operation at a given runway, while taking into account the FDM event severity levels.

Table 3 — example for associating a severity weighting factor with the severity level

Event Severity level	Weighting Factor
Level 0	a
Level 1	b
Level 2	c

Table 4 — example of combining event counts and severity weighting factors

Month	L0 Event Count	L1 Event Count	L2 Event Count	Total score
Jan	30	15	2	$=30*a+15*b+2*c$

### Taking into account the age of the data

The time range on which a measure (e.g. event count, event rate) is computed significantly affects the measure values. For example, a recent change in a trend may not be visible in the measure values if the measure is computed over a long period of time.

## 4. Studying distributions

### Visual assessment of a distribution

This section introduces a few notions to help in studying a distribution of data points. However, before computing any distribution measure, plotting the distribution can be helpful to visually identify some important characteristics and avoid making wrong assumptions, such as assuming that the studied distribution is Gaussian while it is not. For a one-dimensional distribution (i.e. each data point has just one value), plotting the values on a histogram would allow spotting outliers or the presence of several modes, or identifying that the distribution is skewed.

### Types of averaging

Averaging, as a way of describing central tendency can be very useful when trying to summarise a range of values. However, ‘average’ is a concept with multiple definitions that are worth understanding. Before selecting an approach, it is important to understand the type and distribution of data.

- |               |  |
|---------------|--|
| <b>Mean</b>   | This is the most common interpretation of ‘average’ and is defined as the sum of the values divided by the number of values. The mean can only be calculated for numerical (number-based) data. Although it is useful as a central tendency indicator, the mean is easily skewed by extreme values (e.g. from erroneous data).   |
| <b>Median</b> | The median is defined as the central value of a sorted list of data. For this reason, calculating a median value requires the data to be sortable (i.e. apply an order). This implies that the median can be established for both numerical and ordinal data ( <i>Example: assuming that the selected slats/flaps configuration can be {0; 1; 2; 3; FULL}, the median of this list is ‘2’.</i> ). The median is less affected by outliers and skewed data than the mean.     |
| <b>Mode</b>   | The mode is defined as the most commonly appearing value and can also be useful when there is a need to ignore extreme values. However, the mode of a distribution is not necessarily unique; some distributions may have more than one value that appears an equal number of times (bi-modal or multi-modal). The mode can be determined for numerical data, ordinal data, and so-called ‘nominal’ data (data with only a name value, such as <i>destination airport</i> ). |

In a normal distribution (also designated as Gaussian distribution), the mean, median, and mode all have identical values.

### Skewness of a distribution

‘Median’ and ‘mean’ are good central tendency measures that can be easily calculated. Comparison of these values will make us decide on the distribution of the data.

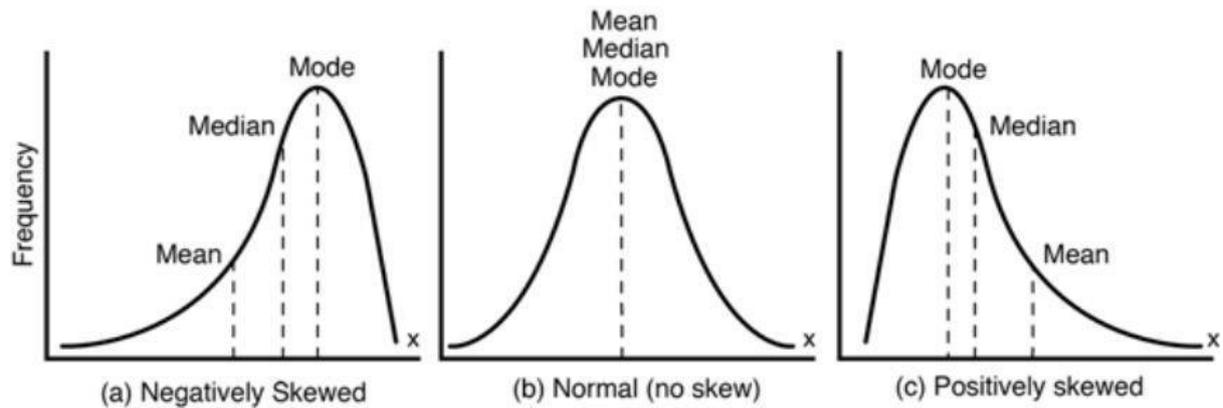
- If Mean < Median < Mode: the distribution is skewed to the left. In this case, there is a negative skew.
- If Mode < Median < Mean: the distribution is skewed to the right. In this case, there is a positive skew.
- If Mean = Median = Mode: the distribution is symmetrical.

If a distribution is skewed, the data should be analysed with complementary techniques and the causes of skewness in the distribution should be investigated.

Although averaging can provide useful information about a range of values, it is still possible for an average measurement to remain unchanged despite a change in the underlying values.

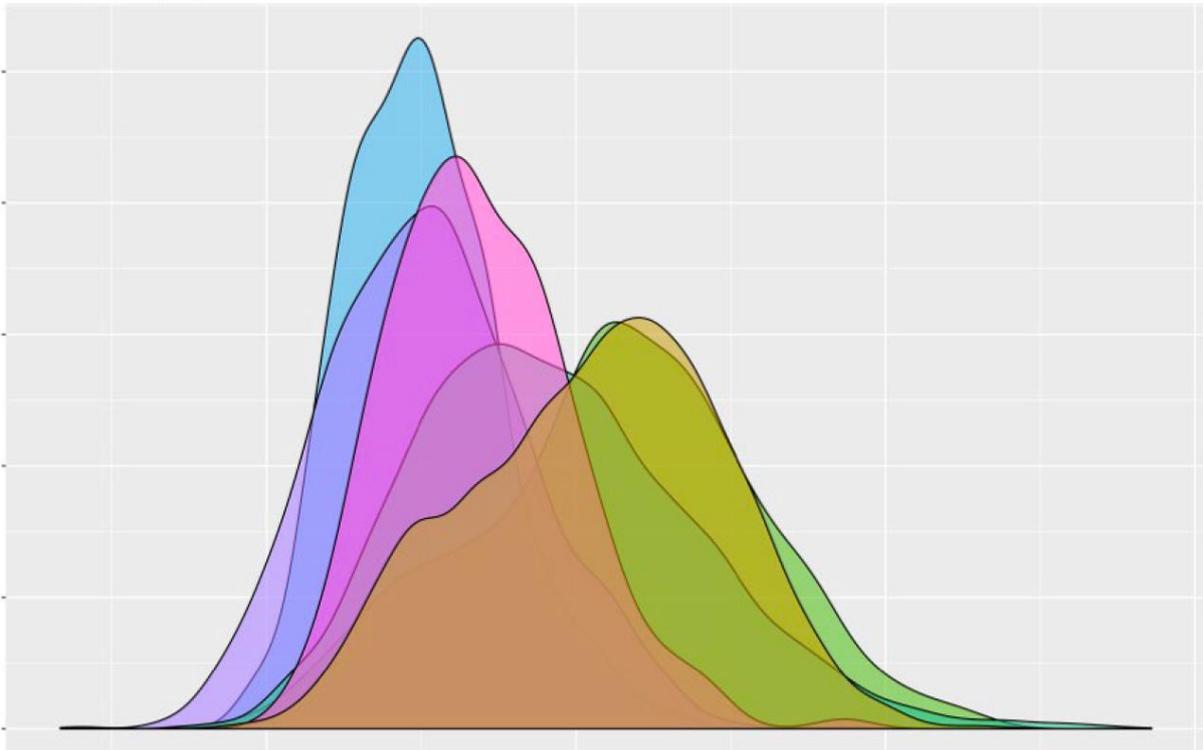
Figure 5 shows examples of curves corresponding to skewed and symmetrical distributions.

Figure 5 — mean, median, mode, and skewness of a distribution



To avoid this, and to better understand the data, it is advisable to plot the actual distribution against comparable data sets. For example, the rotation technique is usually specific to a given aircraft model, therefore, mixing data regarding rotation rates from different aircraft models may not be helpful (see figure 6). Regarding visualisation, see Section III.7.

Figure 6 — density histograms of a flight parameter for different operations



### Standard deviation and quartiles

#### *Standard Deviation*

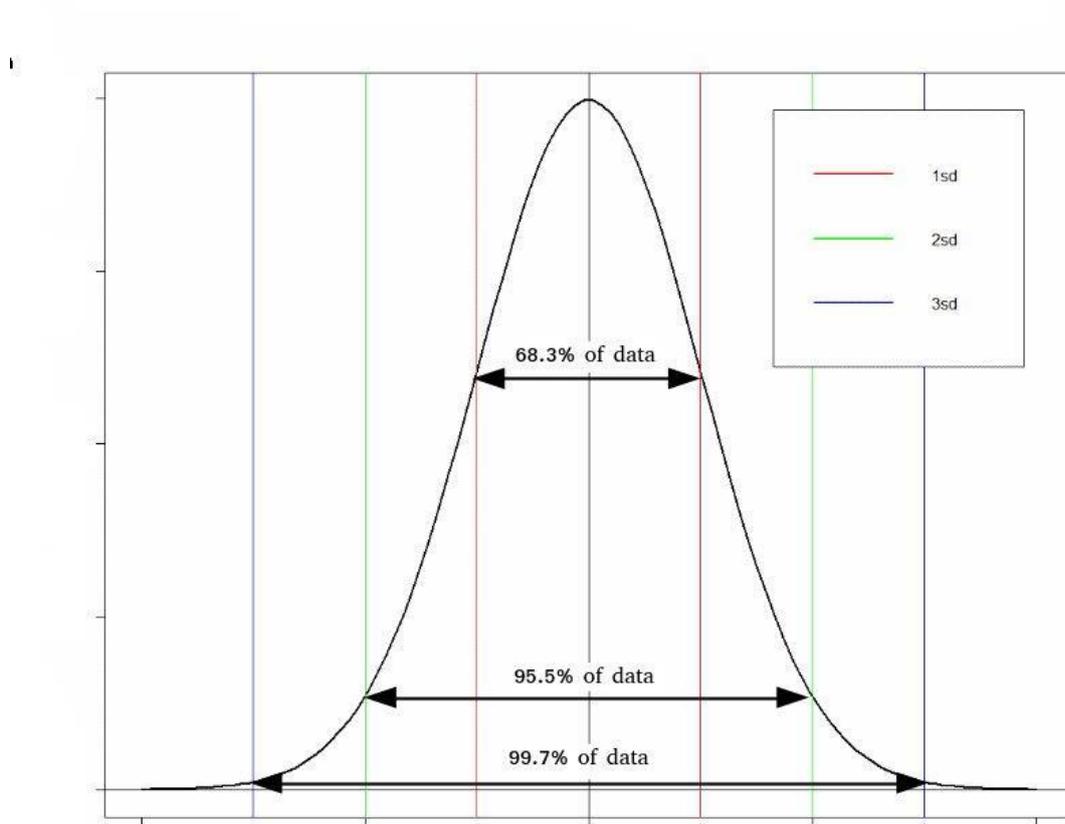
The **standard deviation (SD)** of a sample is used for characterising the dispersion of the sample values from the mean of that sample. It is calculated as shown below:

$$s = \sqrt{\frac{\sum_{i=1}^n (x_i - \mu)^2}{n}}$$

Where:

- $x_i$  designates the values of the sample,
- $n$  is the total number of values in that sample,
- $\mu$  is the mean of all values of the sample:  $\mu = \frac{\sum_{i=1}^n x_i}{n}$ .

Figure 7 – standard deviation of a Gaussian distribution



### Quartiles

**Quartiles** are frequently used for studying a distribution and define the percentage of data above certain defined quantities or tiles. Quartiles, as the name implies, divide the distribution in 4 quarters as follows:

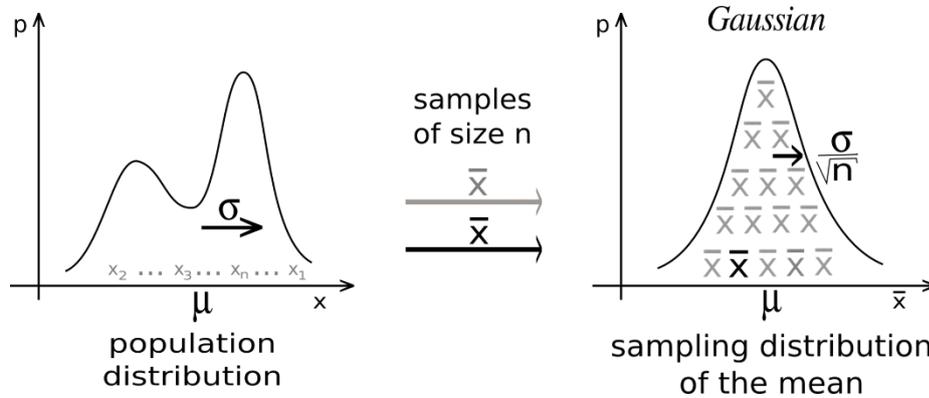
- First quartile Q1 splits the lowest 25 % of the sorted values of the distribution from the highest 75 % of the sorted values.
- Second quartile Q2 (frequently called Median) splits the lowest 50 % of the sorted values from the highest 50 % of the sorted values.
- Third quartile Q3 splits the lowest 75 % of the sorted values from the highest 25 % of the sorted values.
- Interquartile range (IQR) is the difference between Q1 and Q3 values:  $IQR=Q3-Q1$ .

### Central limit theorem, Z-scores, and confidence interval

The central limit theorem (CLT) states that the distribution of the sample means approximates a normal distribution, as the sample size becomes larger. Consequently, the mean of a sample of data trends to the mean of the overall population in question as the sample grows (see Figure 8). Hence,

although the data might not be normally distributed, the CLT allows to perform statistical analyses that would require a normal distribution.

Figure 8 — the central limit theorem (source: Wikipedia)



*Note:*

The CLT relies on some assumptions, for instance that each observation is randomly generated in a way that does not depend on the values of other observations.

Based on this assumption, the **Z-score** of a given value  $x_i$  of a sample is calculated with the formula below:

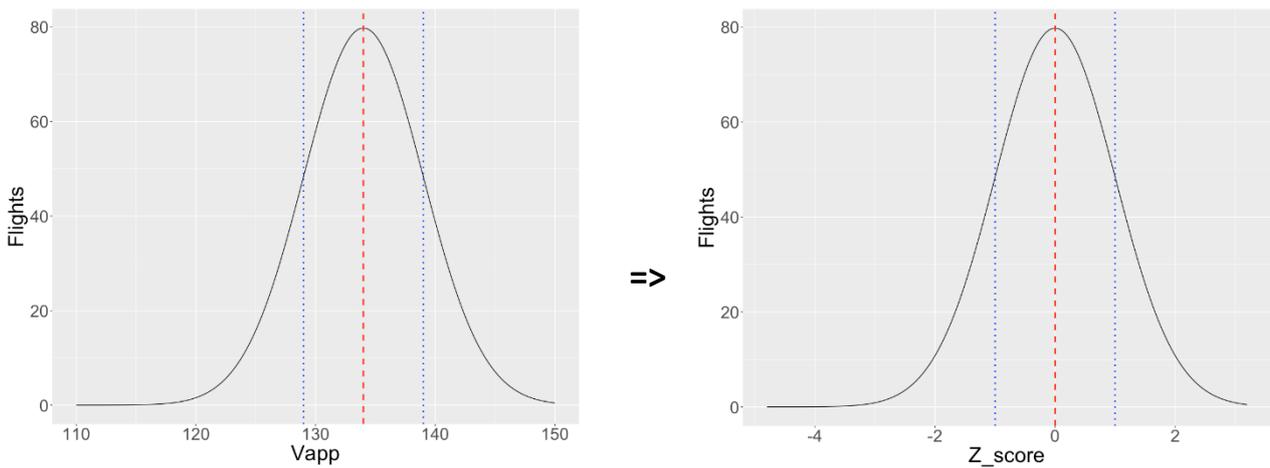
$$z = \frac{x_i - \mu}{s}$$

where:

- $\mu$ : mean of the sample
- $s$ : SD of the sample

This conversion equation corresponds to 'normalising' a gaussian distribution, that is to say, if  $x$  was a normal distribution with mean  $\mu$  and standard deviation  $s$ ,  $z$  would then be the standard normal distribution (mean is 0 and standard deviation is 1). See Figure 9.

Figure 9 — example illustrating the transformation into Z-scores.



In Figure 9, the left-hand plot shows a normal distribution of approach speeds for a given fleet and the right-hand plot shows the standard normal distribution.

The probability value corresponding to the z value is found in mathematical tables called Z-score tables (also called standard normal distribution tables). In particular, the positive Z-score table shows the percentage of values that are below a Z-score value in a standard normal distribution, corresponding to the area below the standard normal distribution and to the left of the Z-score value (see Figure 10).

For example, Table 5 shows that 95 % of values of a standard normal distribution are below 1.65.

*Note:*

Examples of Z-score tables can be found on the internet.

Figure 10 — relation between values in a positive Z-score table and the Z-score

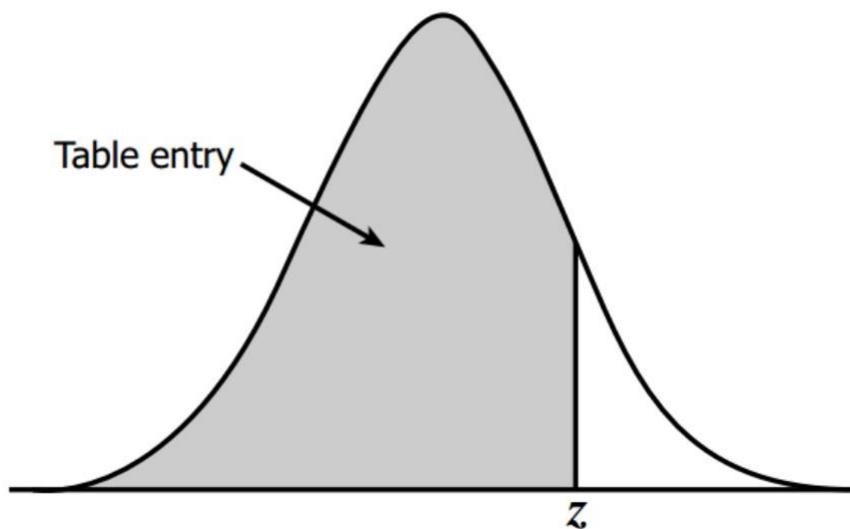


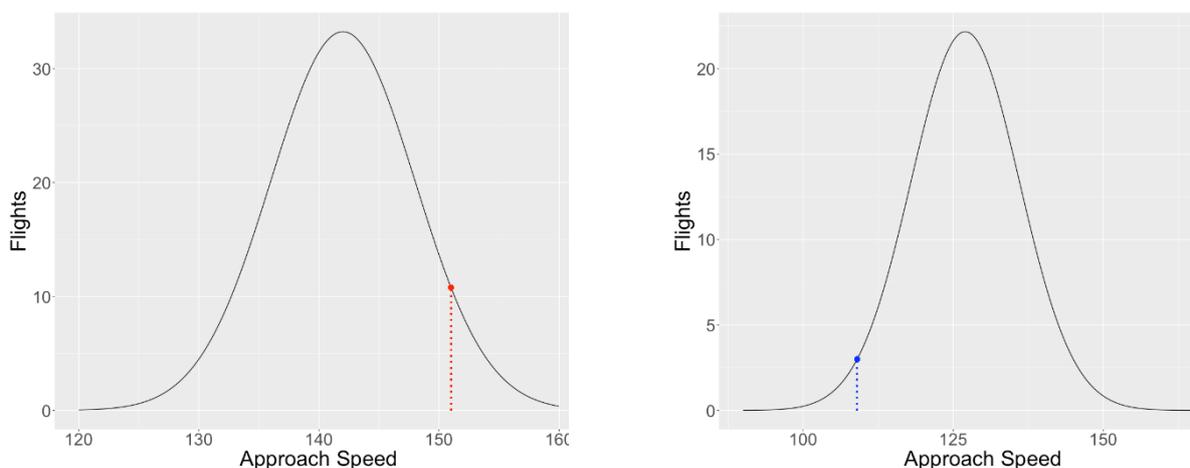
Table 5 — excerpt of a positive Z-score table.

z	0	0.01	0.02	0.03	0.04	0.05	0.06	0.07	0.08	0.09
---	---	------	------	------	------	------	------	------	------	------

0	0.5	0.504	0.508	0.512	...	...	...	...	...	...
0.1	...	...	...	...	...	...	...	...	...	...
0.2	...	...	...	...	...	...	...	...	...	...
...	...	...	...	...	...	...	...	...	...	...
1.5	...	...	...	...	...	...	...	...	0.9429	0.9441
1.6	0.9452	0.9463	0.9474	0.9484	0.9495	<b>0.9505</b>	0.9515	0.9525	0.9535	0.9545
1.7	0.9554	0.9564	...	...	...	...	...	...	...	...
1.8	...	...	...	...	...	...	...	...	...	...
1.9	...	...	...	...	...	...	...	...	...	...

*Example 1:*

Figure 11 — two distributions of the approach speed



Is a value of 151 from the distribution shown in the left-hand plot of Figure 11 (with a mean of 142 and SD of 6) a greater deviation from the mean than a value of 109 from the distribution in the right-hand plot (with a mean of 127 and SD of 9)?

Converting into Z-scores allows to compare these two values.

$$z_1 = \frac{151-142}{6} = 1.5$$

$$z_2 = \frac{109-127}{9} = -2$$

The second value (109) corresponds to a greater deviation because the absolute value of its Z-score is larger.

*Example 2:*

Let's consider flight data corresponding to a year of operations covering all seasons and routes and collected with a low flight capture rate (10 % of flights, for example). The analyst would like to infer the mean for the values of a given flight parameter.

When calculating the confidence interval for the distribution mean, the analyst applies the confidence interval formula as shown below.

Boundaries of the confidence interval =  $\mu \pm (Z * (s/\sqrt{n}))$

Where:

- $\mu$  is the computed mean of the sample
- Z is a chosen Z-score value corresponding to a level of confidence
- s is the computed standard deviation of the sample
- n is the number of observations in the sample

According to Z-score tables:

- The Z-score value corresponding to a level of confidence of 99 % is 2.58.
- The Z-score value corresponding to a level of confidence of 95 % is 1.96.
- The Z-score value corresponding to a level of confidence of 90 % is 1.65

*Note:*

The explanations in this section should not be misunderstood as recommendations to model FDM data samples with the normal distribution law. A normal distribution has 'thin tails', meaning that the probability of extreme values computed from a normal distribution is very low and often it does not match FDM experience. Other types of distributions exist that may be better suited for modelling. For example, the gamma distribution or the generalized extreme value distribution may be more appropriate in some cases.

Example:

An operator wishes to assess the probability of a hard landing with normal acceleration value exceeding 3.0 g. It has never experienced such a hard landing in its operation and therefore the available FDM data do not contain any occurrence of normal acceleration value exceeding 3.0 g at landing.

Assuming that this operator has routinely collected normal acceleration values for all landings, it will then be able to fit these values to a distribution law, thereby producing a model of the data from which the probability of a normal acceleration value exceeding 3.0 g could be determined. If the operator would use a normal distribution to model the distribution of collected normal acceleration values, this may result in grossly underestimating the probability of an extreme value, and in particular, the probability of a normal acceleration value at landing exceeding 3.0 g.

## Analysis of variance

The **analysis of variance (ANOVA)** is used for detecting differences between two or more group means. A typical application of ANOVA is to determine whether it is likely that two samples of values are random samples from the same population, i.e. that the differences observed between the values in the two samples are just caused by chance.

Statistical hypothesis testing involves evaluating a claim made about the value of a distribution parameter. The claim is formed of two parts: the null hypothesis ( $H_0$ ) and the alternative hypothesis ( $H_1$ ).

The null hypothesis is considered true until evidence indicates otherwise.

If it can be concluded that the null hypothesis is false, then the alternative hypothesis must be true since the alternative hypothesis is mutually exclusive to the null hypothesis. However, failure to reject the null hypothesis cannot be taken as proving the null hypothesis, it only means that the alternative hypothesis could not be proven.

There are two potential errors that can be made in hypothesis testing:

Type I error: The error of rejecting the null hypothesis while it is true and should not be rejected.

Type II error: The error of not rejecting the null hypothesis while it is false and should be rejected.

The risk of making these two errors is inversely linked – for a given scenario and a given sample, as the risk of a type I error increases, the risk of a type II decreases. However, it is possible to decrease the risk of a Type II error by increasing the sample size.

There are many applications of ANOVA (one way, two ways, factorial ANOVA, repeated measure ANOVA, MANOVA, etc.).

*Example (one-way ANOVA)*

An example is given below about one-way ANOVA. It is assumed that the FDM analyst wants to analyse the differences between the FDM event count means of 4 runways. 8 of 40 FDM event counts are given for each runway in Table 6.

Table 6 – FDM event counts for four runways

Runway No 1	Runway No 2	Runway No 3	Runway No 4
10	45	40	17
5	20	15	12
24	24	19	31
15	26	21	22
36	12	7	43
25	28	23	32
12	17	12	19
14	26	21	21

First step: FDM analyst set hypothesis.

Null Hypothesis (H0) is that the observed differences in the numbers of event counts between runways are not statistically significant, i.e. these observed differences are not unlikely to have been caused just by chance.

H0:  $\mu_1 = \mu_2 = \mu_3 = \mu_4$  (where  $\mu_i$  is the means of FDM event counts for runway No i)

Alternative Hypothesis (H1 or Ha) is that there is statistical significance, i.e. observed differences are unlikely to have been caused just by chance.

H1: At least one of these runways has a higher or lower number of FDM events.

Second step: one-way ANOVA is applied.

Result of the ANOVA, p - value and F statistics is extracted and H0 accepted or rejected depending on these statistics. If H0 is rejected, at least one of them is different but ANOVA does not indicate which runway is different. For this information, some statistical tests are needed.

*Note:*

The F statistics of a group of data points is the ratio (between-groups variance)/(within-group variance). When H0 is true, the F-statistics is close to one, i.e. between-groups variance is close to within-group variance.

### Case of a discrete variable: the binomial distribution and the Poisson distribution

The Gaussian distribution is not applicable to the case of a discrete variable, such as a flight parameter that can only take a finite number of values (e.g. flap settings, TAWS warnings), or a finite number of possible outcomes during a flight.

If the variable can only take two values (for instance, 'TCAS RA' or 'no TCAS RA'), then one usually refers to a **binomial distribution**. When considering a binomial distribution, if the probability of taking one of these two values during one 'trial' (i.e. a flight, or a day of operation) is known, then the probability of taking r times this value in n trials is provided by the following formula:

$$P(r)_n = \frac{n!}{r!(n-r)!} p^r (1-p)^{n-r}$$

Where:

- p: probability of a 'success' in 1 trial
- n: number of trials
- P(r)<sub>n</sub>: probability of r successes in n trials

The **Poisson distribution** is a discrete probability distribution that expresses the probability of a given number of events occurring in a fixed interval of time or space (if these events occur with a known constant mean rate and independently of the time since the last event)<sup>23</sup>.

One benefit of the Poisson Distribution is that calculating the probability of a certain number of events occurring in a certain period requires knowledge of only the existing mean (μ) (e.g. the mean value of a given occurrence per 1 000 sectors and per month).

The probability of x events occurring in a period [written as p(x) ] can be expressed as:

$$p(x) = \frac{e^{-\mu} \mu^x}{x!}$$

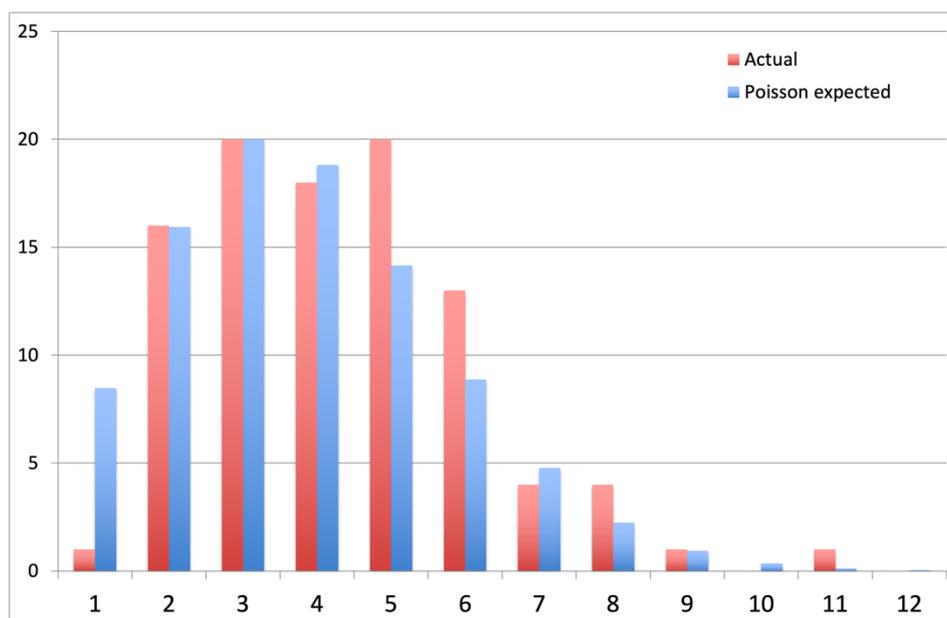
where e is Euler's number (2.718...), μ is the mean and ! is the factorial function.

---

<sup>23</sup> The Poisson distribution can be derived as a limiting case of the binomial distribution taken as the number of trials goes to infinity and the expected number of successes remains fixed.

Comparing the number of events observed over the latest period with the likelihood expressed by the Poisson Distribution can help guide whether this result was likely or unlikely and can therefore help to inform what action is necessary. See also Figure 12.

Figure 12 — distribution of measured monthly event rates versus the expected monthly rates calculated with the Poisson distribution equation, using the mean monthly value



## 5. Dealing with outliers

An outlier can be defined as an observation (or group of observations) which seems not to be consistent with the set of data. It is an easy idea, which needs to be accurately addressed for each case of study because it requires to know the distribution of the numerical variables.

### General

#### *What is an outlier?*

#### Sentinel values

In some cases, detection of outliers corresponds to so-called 'sentinel values'. These values are used to represent a particular situation in a numerical variable. For example, a sensor could use a certain number out of its common range of values to represent that it is powered or not. Analysts working with data sets can as well introduce sentinel values to represent a condition (e.g. 'not applicable' or 'unknown' values will be encoded as 99999).

#### Definition of an outlier

According to the NIST/SEMATECH e-Handbook of Statistical Methods published by US National Institute of Standards and Technology, 'An outlier is an observation that lies an abnormal distance from other values in a random sample from a population. In a sense, this definition leaves it up to the analyst (or a consensus process) to decide what will be considered abnormal.' Therefore, before outliers can be singled out, it is necessary to characterise normal observations.

### *Importance of addressing outliers*

Performing an analysis with multiple outliers can result in a false outcome. However, the process of dealing with outliers needs special attention because, if not correctly done, it might have a significant impact on the analysis. There are multiple techniques to deal with outliers depending on how the data looks like, explaining some of them would help.

Finally, an outlier might be a real value and some knowledge of the order of magnitude and/or sensor limits is needed to distinguish between a real value, or an outlier caused by other reasons (e.g. sensor reading error, integration of data, etc). If the outlier corresponds to a real event, then a corrective action may need to be done urgently, e.g. a technical inspection after a hard landing.

### Detection and assessment of outliers

#### *Use of domain knowledge*

Sometimes, the typical range of a value is known.

For example, when measuring the normal acceleration at landing, a typical range could be between 1 and 2 g. Therefore, normal accelerations values outside of the domain [1 g; 2 g] could be considered abnormal, whether they are outliers or not.

Another example is the typical airspeed during approach, for which the reference speed ( $V_{ref}$ ) in the flight crew operating manual provides indicative value.

#### *Comparing the mean and the median*

A histogram can be a helpful visual tool for a quick overlook of a distribution. However, the presence of outliers can be determined even without plotting data, by comparing the median and the mean of a distribution. The examples below illustrate the fact that the median is much less affected by outliers in a set of values than the mean. This feature may be used as a quick check of the data to identify outliers.

#### Example 1: Test data without outliers

A test set of 100 values was randomly generated by a function simulating a normal distribution with a mean of 10 and a standard deviation of 1. This test data is displayed in a histogram (see Figure 13) and in the below summary (see Table 7).

Figure 13 — example of distribution of a randomly generated sample

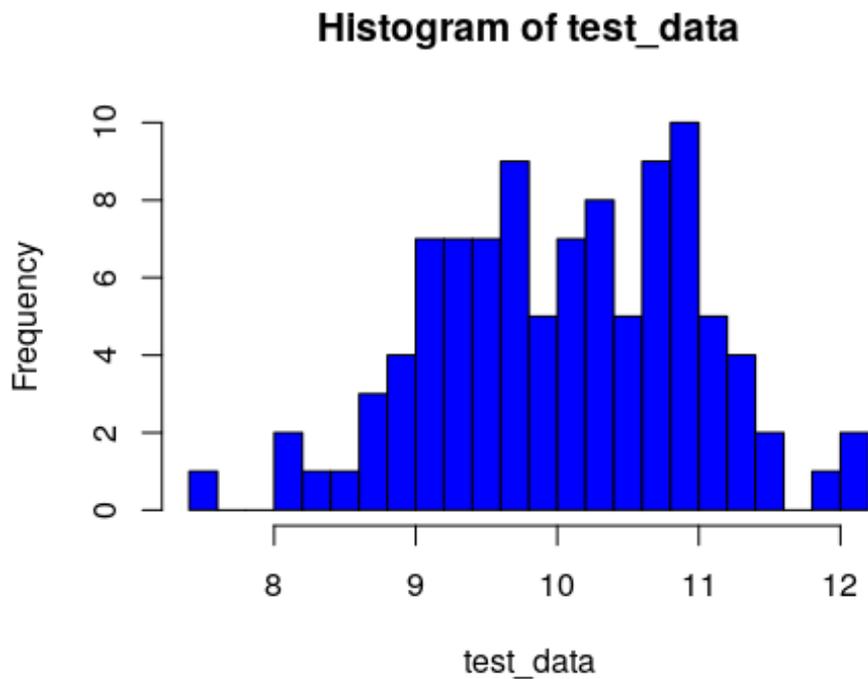


Table 7 — Summary of the test data presented on Figure 13

Minimum	1 <sup>st</sup> Quartile	Median	Mean	3 <sup>rd</sup> Quartile	Maximum
7.593	9.376	<b>10.065</b>	<b>10.046</b>	10.758	12.101

Example 2: Test data with outliers

In the same set of 100 values, two values of 1034 and 1054 well beyond the range of values from the initial set were inserted to simulate two outliers. This data is displayed in a histogram and in the below summary (see Figure 14 and Table 8).

Figure 14 — example of of distribution of a randomly generated sample, in which two outlier values were inserted

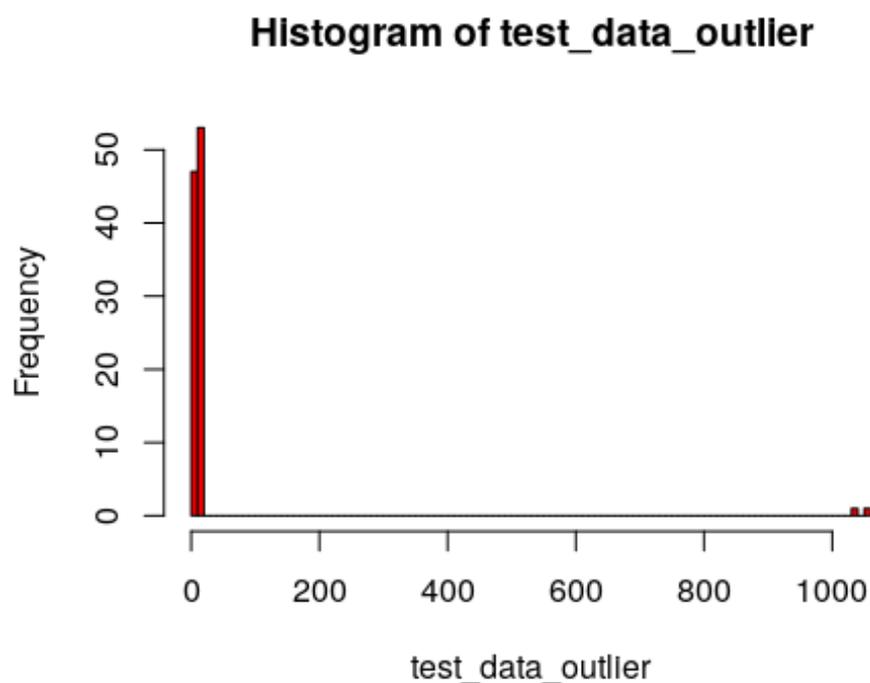


Table 8 — summary of the test data presented on Figure 14

Minimum	1 <sup>st</sup> Quartile	Median	Mean	3 <sup>rd</sup> Quartile	Maximum
7.593	9.386	<b>10.069</b>	<b>30.319</b>	10.831	1054

Examining the summary of the data in Table 8, the first observation is that the mean and the median values are not of the same magnitude and the second is that the median value was only slightly affected by the outliers when comparing to the median values obtained from the previous example. The mean value was significantly affected by introducing outliers in the dataset.

To conclude, a summary of the data containing the mean and the median can be very helpful to identify that outliers may be present in the sample.

#### *Using boxplots*

When some outliers have been detected, two questions may come to mind. Is the outlier value stemming from a data quality issue or a programming error, or is it genuine? If it is a genuine value, the analyst should analyse this outlier value.

For detecting outlier values, a boxplot could also be used.

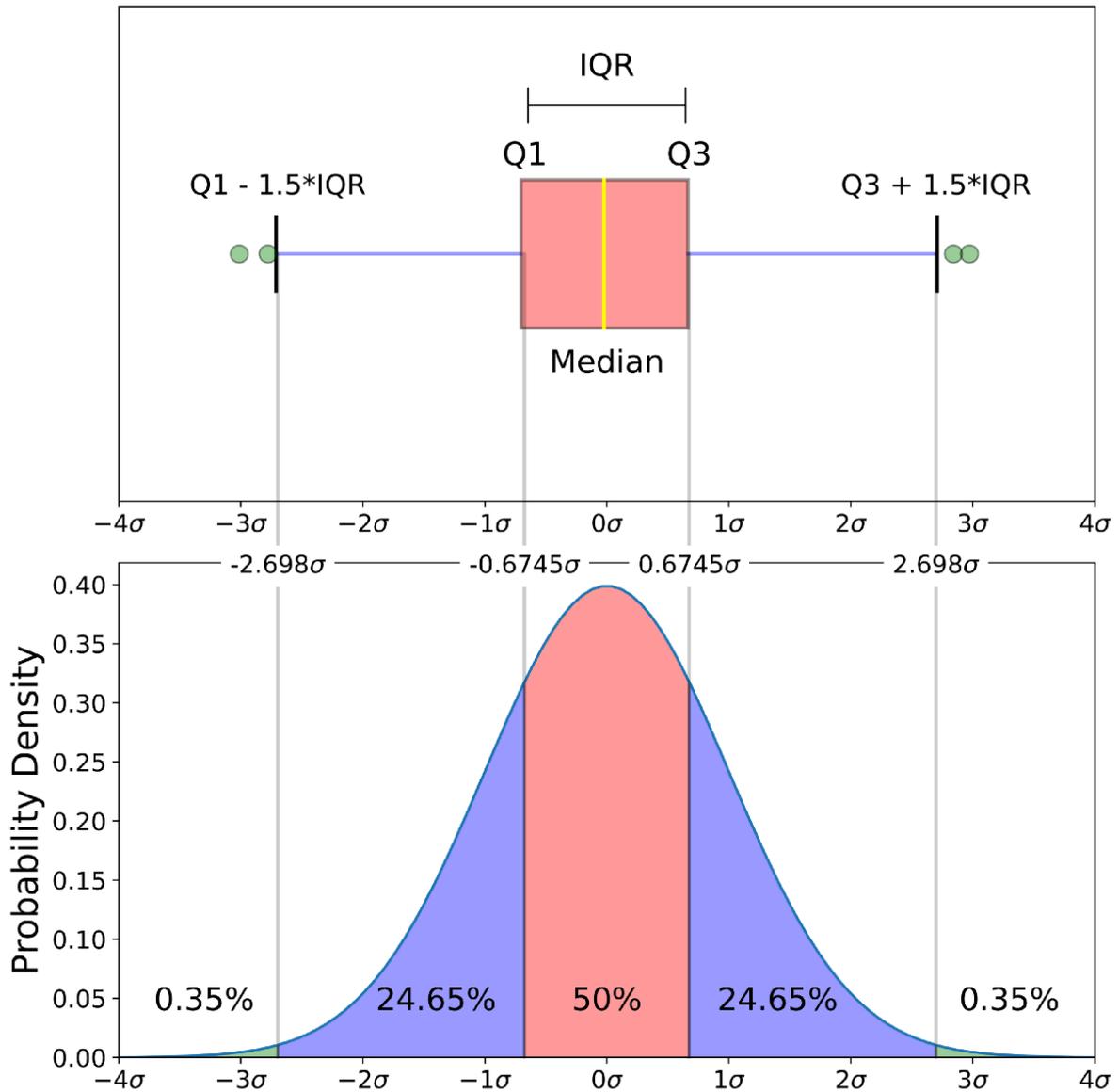
According to the NIST/SEMATECH e-Handbook of Statistical Methods, a boxplot is 'constructed by drawing a box between the upper and lower quartiles with a solid line drawn across the box to locate the median. The following quantities (called fences) are needed for identifying extreme values in the tails of the distribution:

- lower inner fence:  $Q1 - 1.5 \cdot IQ$
- upper inner fence:  $Q3 + 1.5 \cdot IQ$
- lower outer fence:  $Q1 - 3 \cdot IQ$
- upper outer fence:  $Q3 + 3 \cdot IQ$

In addition, the NIST/SEMATECH e-Handbook of Statistical Methods proposes that 'a point beyond an inner fence on either side is considered a mild outlier. A point beyond an outer fence is considered an extreme outlier.'

As shown in Figure 15, in the case of a normal distribution, 99.3 % of the values are between the lower and inner fences, i.e. 0.7 % of the values would be considered outliers according to the NIST/SEMATECH e-Handbook of Statistical Methods.

Figure 15 — boxplot of a normal distribution (Author: Michael Galarnyk, source: towards data science)



Example of use of boxplots

Consider that the event rates of all FDM event algorithms are compared to each other using a boxplot chart. This would allow visually identify FDM event algorithms with a non-typical FDM event rate.

Figure 16 shows a boxplot example for an operator. In this example FDM event rates (y axis – values in %) based on the severity level have been compared to each other for different flight cycles and different FDM event algorithms. Every red dot represents an individual outlier FDM event rate for the selected severity and the number of flights considered.

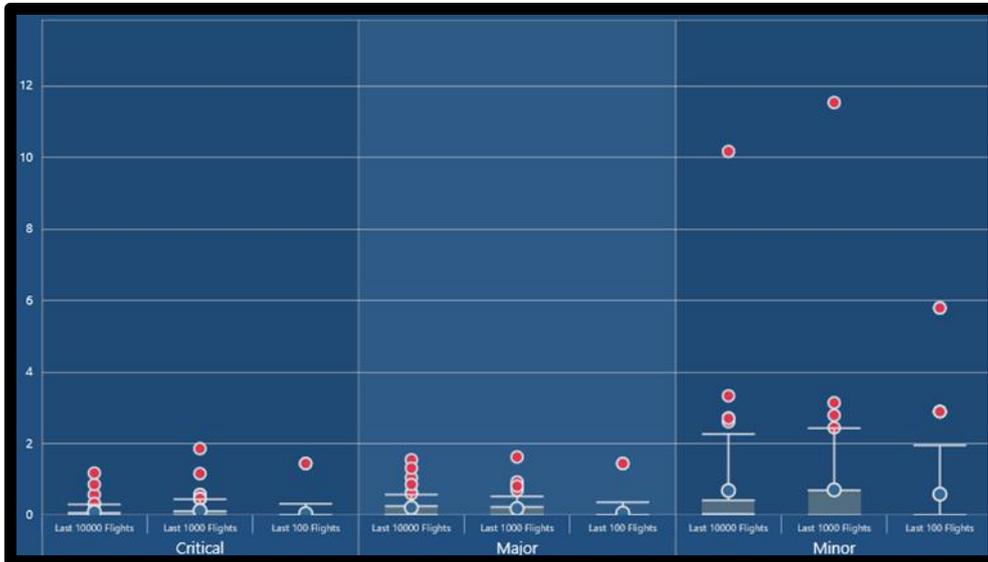
Figure 16 gives an overall picture of those FDM event algorithms that seem to have a much higher rate. This could be used for:

- monitoring any emerging trend in the FDM event set;

- calibrating/standardising the severity levels of FDM event algorithms (in view of creating a composite index);
- benchmarking against other operators that would use the same FDM event algorithms.

While dealing FDM event count and FDM event rate cannot have negative values, therefore lower boundaries are statistically not represented.

Figure 16 — an operator example for using boxplot



To illustrate the boxplot methodology, Figures 17 and 18 below might be helpful. Figure 18 shows 20 data points lined up on a blank numerical axis and illustrates the terms that are used for calculations of boundaries.

Figure 17 — factors of boxplot

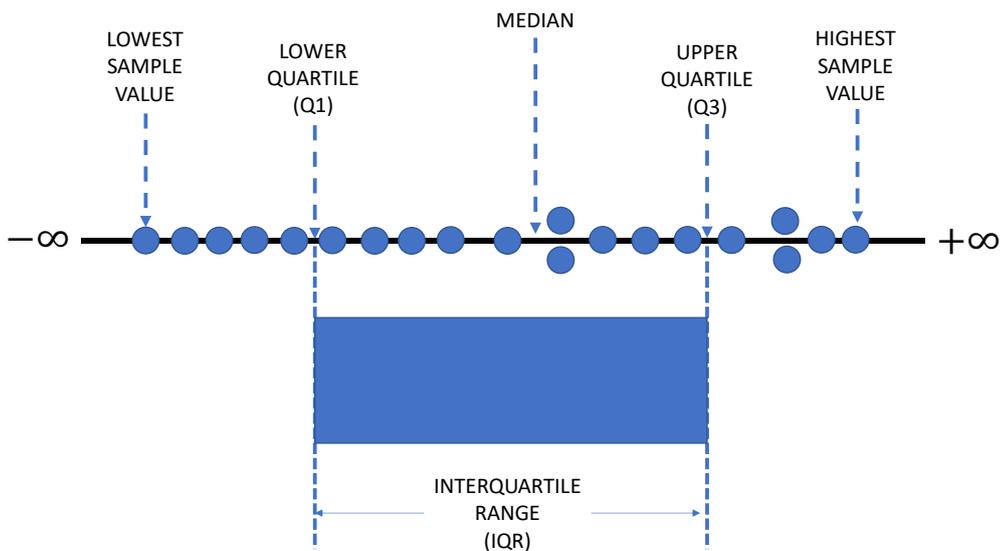
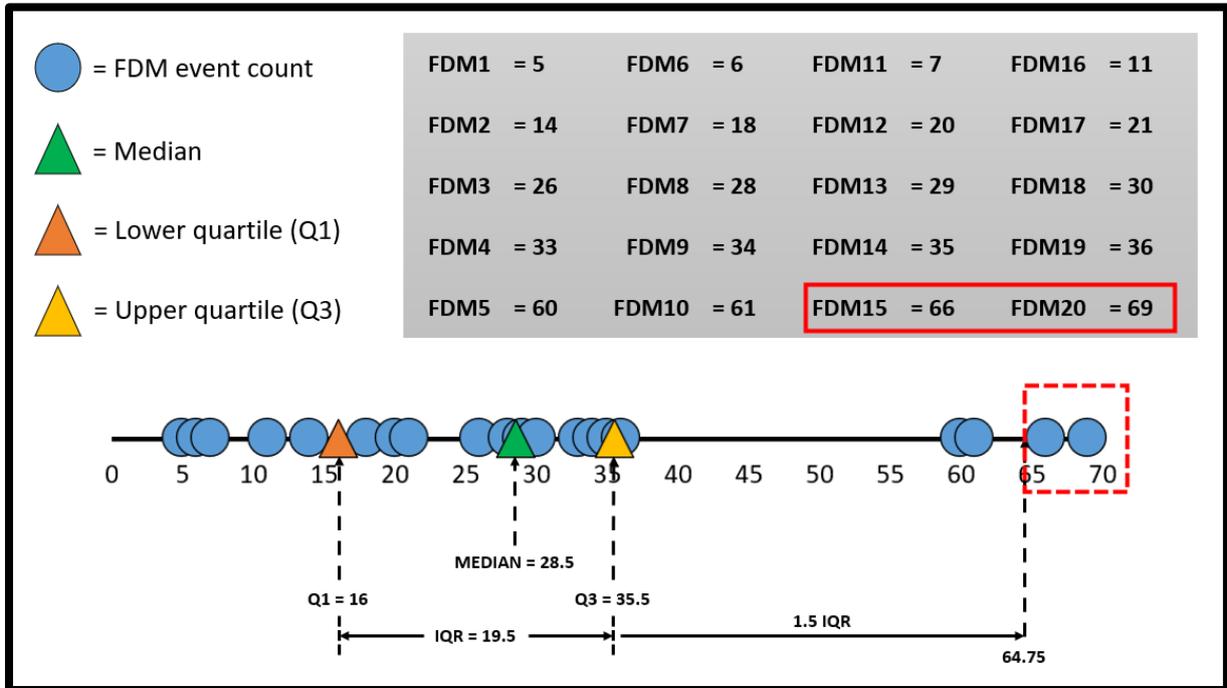


Figure 18 shows an example of use of the boxplot formula to calculate boundaries and identify the outliers. In this example, the two FDM event algorithms No 15 and No 20 are defined as outliers due to their high number of event counts.

Figure 18 — example of using boxplot to identify FDM event counts that seem to be outliers



**Note:**

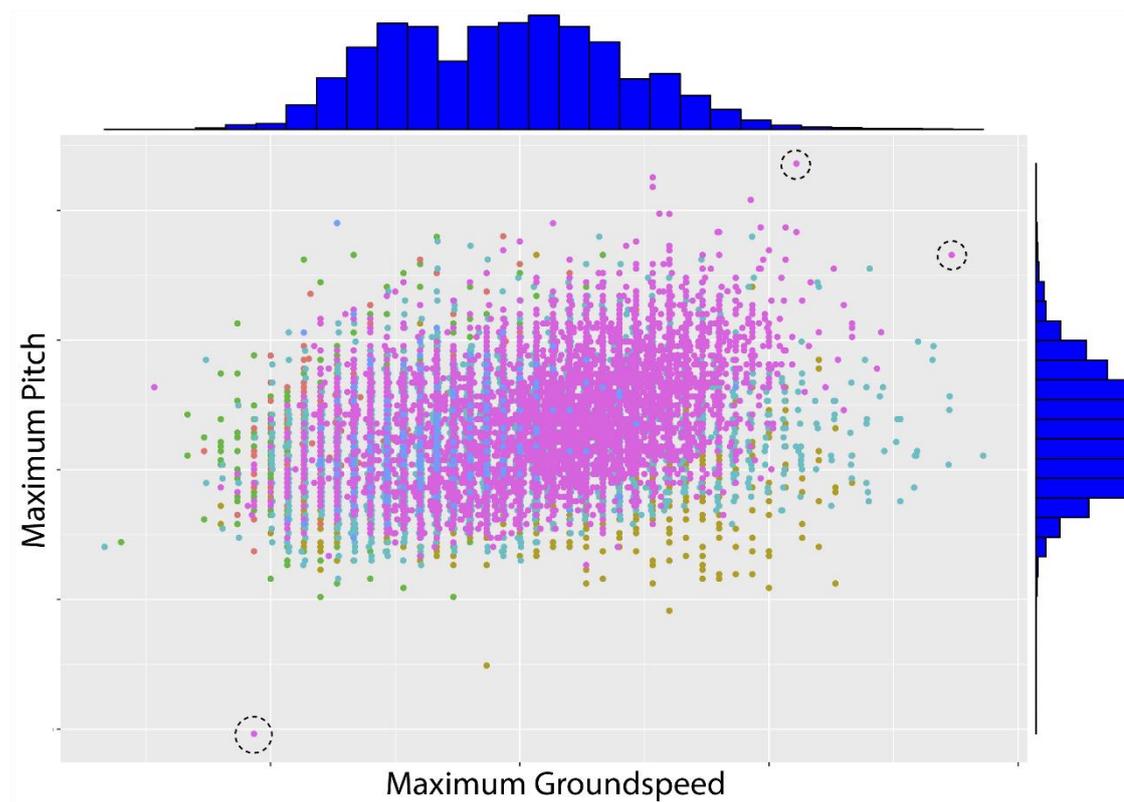
When working with FDM event counts and rates there is no need to calculate the lower boundaries. However, if the boxplot method is applied to a recorded or calculated FDM parameter such as ‘Airspeed at Touchdown’ or ‘Landing Distance,’ then lower boundaries should be considered as well to see the lower value outliers.

*Using scatter plots*

Figure 19 illustrates the benefit potentially brought by a scatter plot for the visual identification of outliers. Each point of the scatter plot corresponds to a data snapshot that consists of the maximum groundspeed value and the maximum pitch attitude value that were collected during the approach phase of an individual flight. The point colour is used to identify the aircraft operator.

In Figure 19, at least three data snapshots can be visually identified as outliers because their maximum pitch attitude value and/or maximum groundspeed bring them far away from the rest of the points on the scatter plot. The histogram shows that these outliers cannot be visually identified when the distributions of max groundspeed or max pitch attitude of these data snapshots are plotted separately.

Figure 19 — example of a scatter plot. Outliers are circled.



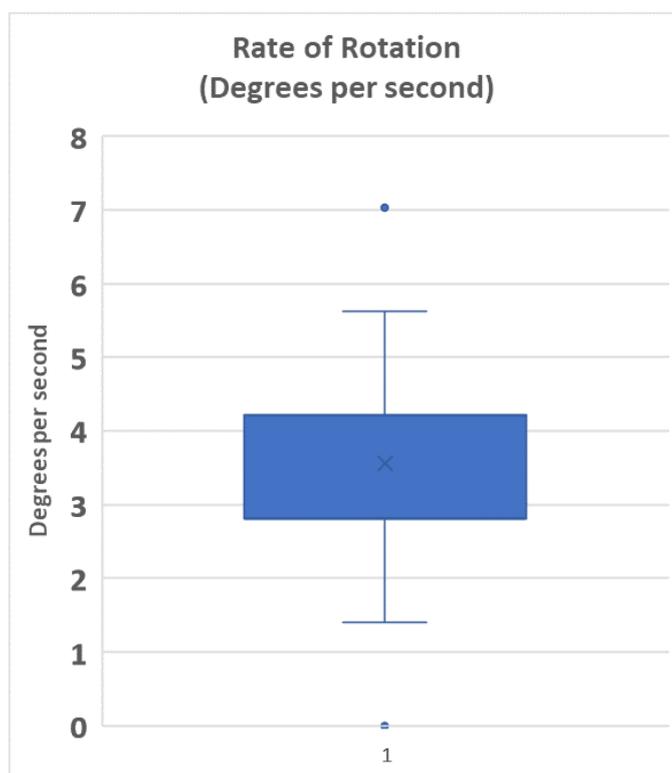
*Looking at related data*

If outliers cannot be immediately discarded, for instance because their value is possible, then further evaluation might be needed. It can be beneficial to include related data. This is called multivariate outlier condition and designates a combination of values that in one dimension seem reasonable but when combined may be unusual.

Example:

The boxplot in Figure 20 shows a boxplot of the rate of rotation during take-off, from a sample of approximately 1 100 individual flights.

Figure 20 — boxplot, rate of rotation outliers are indicated by blue points



In this example, additional data was gathered:

- Departure airport (determined by correlating the latitude and longitude at the beginning of the flight with an airfield database);
- Departure runway (determined by using the heading and latitude, longitude against a database of runways);
- Indicated airspeed at rotation point; and
- Max pitch attitude achieved at rotation.

For each of the flights showing '0' for the rate of rotation at take-off, the related data was checked: see Table 9.

Table 9 — values obtained from the flights where the computed value of the rate of rotation was 0

Rate of Rotation	IAS @ Vr	Max pitch @ VR	T/O Runway	Departure Airport
0	NULL	NULL	NULL	NULL
0	NULL	NULL	NULL	NULL
0	NULL	NULL	NULL	NULL
0	NULL	NULL	NULL	NULL
0	NULL	NULL	NULL	NULL

As each of the comparable data parameters have a NULL value, it can be concluded that the recording of the flights where the rate of rotation of '0' are probably corrupt, which was confirmed by further investigation of those flights.

The remaining outlier values appear to have valid values for the related data parameters: see Table 10.

Table 10 – values obtained from the flights

Rate of Rotation	IAS @ Vr	Max pitch @ VR	T/O Runway	Departure Airport
7	112	15.1	07	XXX
7	112	14.8	14	YYY

## 6. Monitoring trends with statistics

Statistics should include areas where the safety risk has been successfully reduced, not only the problematic areas. This can contribute to encouraging efforts of flight crew members and other staff members.

### Trend Evaluation Areas

#### *General Evaluation*

Overall assessment is important to see the full picture. Details should be avoided at this stage. The aim is to identify general trends. On the other hand, this method can also be used to see the results of past changes (changes in training method, newly recruited pilot groups, procedure changes, etc.).

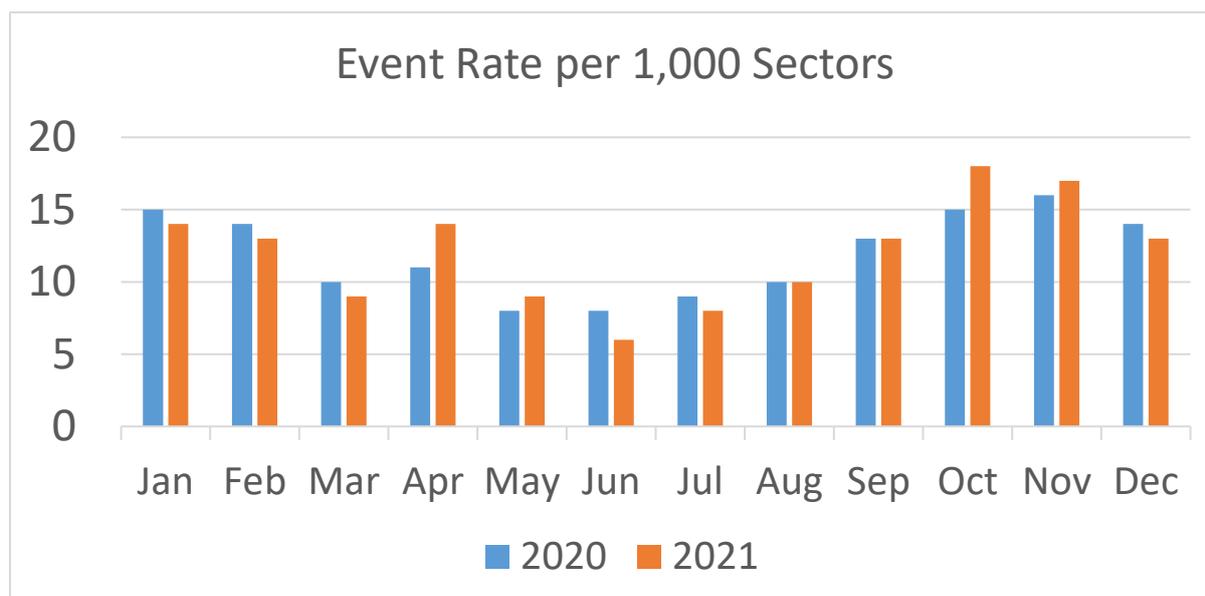
It may be considered to regularly share the general evaluation results with internal stakeholders (such as FDM Board members, flight operations, training departments).

The overall assessment will show where to focus (negative trend in unstable approaches, increase in hard landings, etc.). With detailed evaluations to be made after that, it is necessary to try to find answers to more detailed questions such as when (which phase), where (which runway), how (which event), etc.

For example, an FDM event rate may be tracked over time to look for any developing trend, which can be used for prediction. A simple time series can help to highlight any seasonality in the data.

When there is enough data, it can be useful to make year-on-year comparisons of the time series as shown in Figure 21.

Figure 21 — year-on-year comparison of FDM event rate



### Detailed Evaluations

For detailed evaluations, the most common parameters considered are:

- FDM Event:**  
 Basically, it is the classification and analysis of flight data on the basis of FDM events. It is more flexible and more comprehensive than runway-based analysis.  
  
 This analysis will give a good insight into the behavior and technique of the pilots and the fleet (e.g. late configuration, rush lift-off, etc.). It is an indicator of compliance with SOP.
- City pair:**  
 Specific flight legs might be more prone to events than others (e.g. free seating on the second leg of a triangle flight might lead to large trim changes).
- Destination:**  
 Ranking destinations by event rate can be useful to identify trends linked to the characteristics of a particular airport (e.g. high terrain, steep approach, specific wind conditions, late turns in approach, etc.).
- Runway:**  
 Each runway can carry unique conditions (length, width, terrain, wind, windshear, glide slope, etc.). For this reason, while performing an airport-level analysis is sufficient in the 'General Evaluation' stage, an analysis should be made for each runway of the airport in the 'Detailed Evaluations' stage. This analysis, in a very simple sense, means analysing the data by classifying them based on runways. In this way, it may be possible to take runway specific measures.

Key outputs from this analysis can be:

- to help determine where the risk is;
- to increase the awareness of the pilots about the runway; and

- to create a recommendation for pilots who will fly for the first time on the respective runway.
  
- **Flight crews:**  
Analyses focussed on groups of, or individual, flight crew members are possible. However, the identity of pilots should be protected (see the general note on page 3 of this document). EOFDM document ‘Breaking the silos’, Chapter II addresses data confidentiality aspects and provides examples.  
  
In addition, pilot individual performance monitoring based on FDM may easily lead to situations that are undesirable for the operator, if not carefully implemented. Refer to Section II.3 of EOFDM document ‘Breaking the silos’.
  
- **Other filters:**
  - Aircraft type;
  - Period (e.g. 5-years, year, season, month, week, day);
  - Time window (e.g. time of day, lighting conditions);
  - Operation type (cargo flights, passenger flights, charter vs. scheduled operations, training flights, ferry flights, maintenance check flights, etc.);
  - Area of operations.

#### *Parameters to be Considered:*

- ✓ **Event count**  
Refer to Section III.3.
  
- ✓ **Event rate**  
Refer to Section III.3
  
- ✓ **Event severity**  
Refer to Section II.1.

#### Methodologies

##### *Safety performance indicators (SPIs)*

Before designing FDM-based SPIs, it is important to first identify in the framework of the SMS what needs to be monitored. Whatever the SPI, it brings benefit if it addresses a need identified by the SMS. For example, the need for an SPI may result from a specific risk assessment. Or a SPI may be justified by a decision to monitor precursors to some operational issues that are considered relevant for all operators of a given category of aircraft or for all operations of a given type (e.g. identified in the European Plan for Aviation Safety (EPAS), by a recommendation in an accident investigation report, etc). Then the FDM specialist may be involved to provide FDM data for a specific SPI.

Finally, SPIs are often looking at changes in event rate, location, etc. However important this may be, the real value is analysing operations from a risk perspective i.e. what do these figures and trends actually tell about the risk? This applies equally to FDM-based SPIs as to other FDM programme output. It is thus important to not confuse the monitoring of data trends provided by FDM-based SPIs with actual safety risk trends.

The methods mentioned below reflect the experience drawn from daily practical applications and are just advisory. The content of this section is not exhaustive and other methods can be followed.

#### *Shewhart or control chart*

This method is based on determining a target and trigger levels for the next period, based on the historical data. It relies on the population standard deviation principle, as described in ICAO doc 9859 (safety management manual), Section 4.4.

The average of historical data (for example 3 years) is used to determine the target value. This average of historical data can either be considered as a target value, or the target value can be determined by factoring in expected improvement (for instance, X % lower/higher than the average of historical data), which should be based on an assessment by a subject-matter expert.

After determining the target value, the standard deviation (SD) is calculated for the same set of historical data. Trigger levels are determined by adding the calculated SD to the target value. For example, three trigger levels are defined at 1 SD, 2SD and 3 SD:

TL1= Target value + 1 SD

TL2= Target value + 2 SD

TL3= Target value + 3 SD

#### *Note:*

About 68 % of values drawn from a normal distribution are within 1 SD away from the mean, about 95 % of the values lie within 2 SD, and about 99.7 % lie within 3 SD.

The current data points are then compared with these trigger levels. Depending on how many data points exceed a given trigger level, necessary measures are expected.

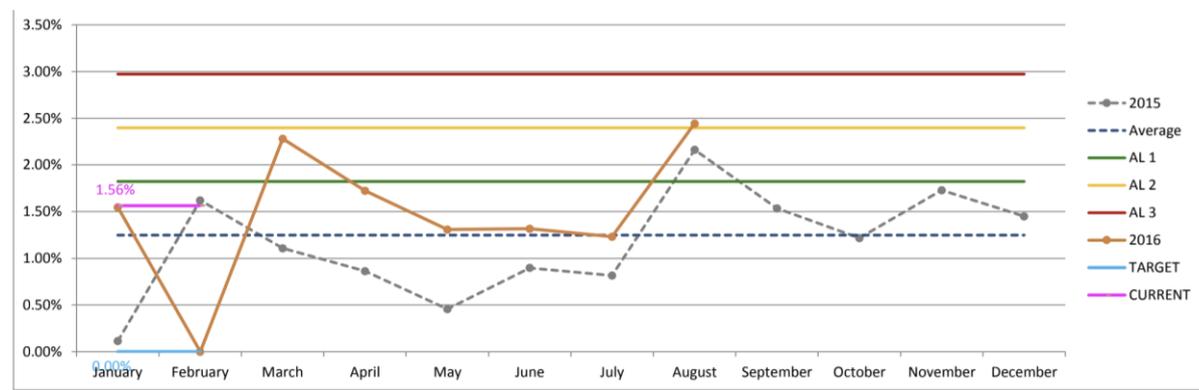
For example: action is due, when a single data point deviates from more than 3 SD from the target value (exceeds TL3), or when 2 consecutive data points deviates from more than 2 SD from the target value (exceed TL2). This would typically trigger further analysis to determine the source and root cause of this trend.

The target value and the trigger levels should be re-computed for each new monitoring period, based on the average and the SD of the previous monitoring period, as applicable.

An illustration of this method is shown on Figure 22.

It should be noted that this method works best for large amounts of data, and it is not so relevant for small data sets.

Figure 22 — example of Shewhart chart for the number of landings outside the touch down zone. Trigger level 1 is indicated by the green line, trigger level 2 by the yellow line, and trigger level 3 by the red line



### Moving average method

The moving average is a simple technique, and it is therefore among the most common tool used by data analysts. It is used to calculate the average of values from a given variable on a given interval. Within the scope of FDM, a time interval (e.g. last X days) can be used as well as a certain number of flights (last 10, 100, 200 flights, etc.).

The reason for the term ‘moving average’ is because the average is calculated according to a position in the values series from the considered variable. As this position changes, the average moves up or down depending on the values considered for the average computation.

The moving average is useful for tracking trends. It shows where to focus when performing flight data analysis. An important advantage of working with the moving average can be explained as follows: a series of single data points show many variations that make the decision-making process difficult. By calculating the moving average, these variations are smoothed out, so it becomes easier to see the general trend.

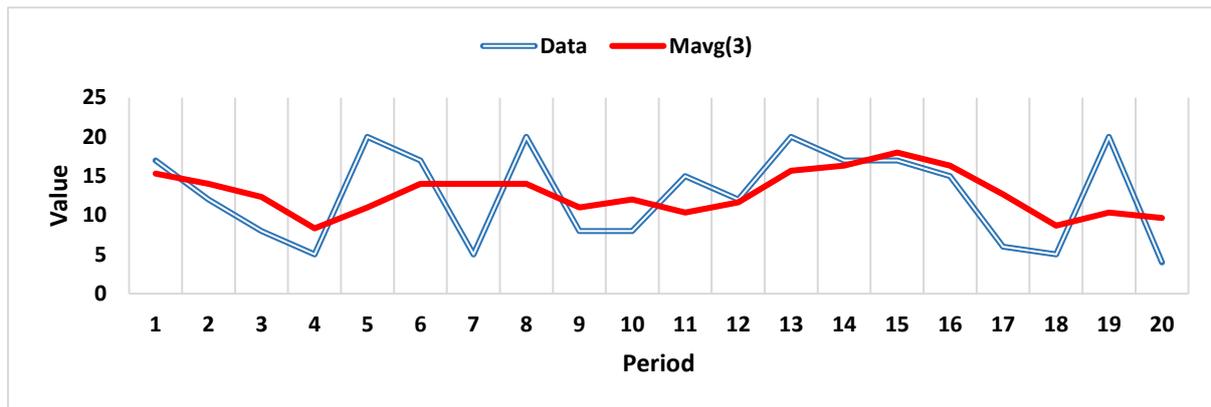
However, the moving average does not give any idea about the reasons for the change in the trend. Therefore, once the general trend has been identified, further analysis should be conducted.

The moving average can be applied in two ways:

### **Comparison of the non-averaged data with averaged data**

In Figure 23, non-averaged data and the 3-day moving average of the same data are plotted together. When the non-averaged data cuts the averaged data up, it means an upward trend. If the non-averaged data cuts the averaged data down, it means a downward trend.

Figure 23 — time series including a moving average

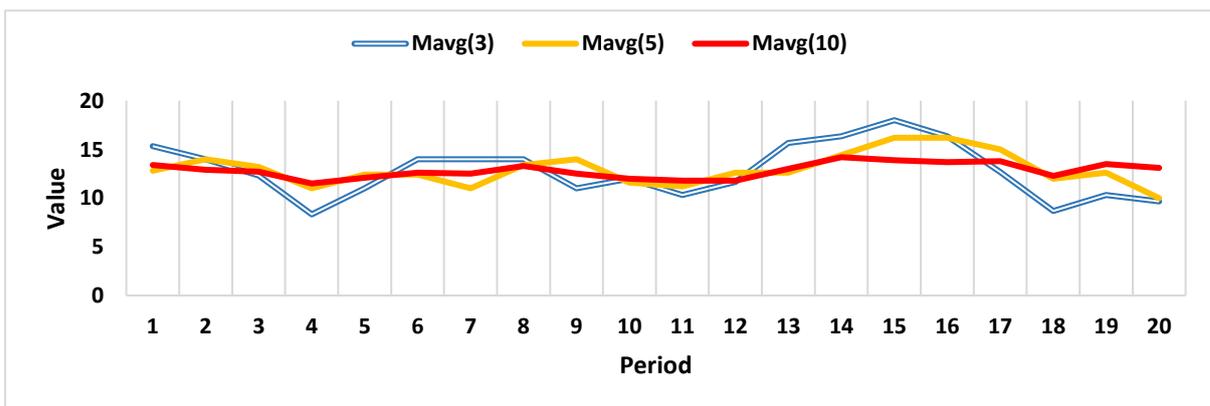


**Comparison between moving averages made on different periods**

The moving averages on different periods are shown on the same graph. Depending on the relative positions taken by these different moving averages, a trend can be identified. In Figure 24, the moving averages over 3, 5, 10 days of the same data are plotted together. If the curve of the moving average with the shortest period (Mavg-3 in the example) is below the curves of the other moving averages (in the example; Mavg-5, Mavg-10), it means a downward trend. On the contrary, if the curve of the moving average with the shortest period is above the other curves, this means an upward trend. If the shortest period is between the other two periods, there is a possibility of trend change.

The size of the intervals considered for computing a moving average will vary according to the scope of the analysis to be made. For example, if long term trend analysis is made on 3 years of data, 180-, 90-, 30-day periods can be used. However, if a shorter-term trend analysis is made on 6-month data, 90, 30, 10 days periods can be used.

Figure 24— plot with three moving averages computed from the same time series, over 3, 5, and 10 points



*Blind benchmarking*

This consists in presenting individual results compared with a global average.

Refer also to EOFDM document ‘Breaking the silos’, Section I.3.

## 7. How to present data

### Preparing data for internal customers

It is important to set up a specific location (folder, webpage, etc.) and to provide contextual information for the ‘internal customers’ of the FDM programme, to help them understand how things are measured or calculated.

In addition, the following questions should be addressed before designing any visual presentation of FDM results:

- **Who are the internal customers?**  
The flight crew training department, the flight operations department, flight crews, the maintenance department, etc., could benefit from the output of the FDM programme. For more information about potential use of the FDM programme beyond the SMS and day-to-day use of flight data, refer to Chapter III of EOFDM document ‘Breaking the silos’.
- **Which data can internal customers access?**  
A system should be designed to take into account the privacy and security of flight data. All aspects of data sharing should be documented. Refer to Chapter II of EOFDM document ‘Breaking the silos’.
- **Combining FDM outputs with other results?**  
Thanks to the developments in data processing technologies, it is now possible to automatically combine FDM events and other outputs of the FDM programme with other data sources (ASR, weather data, traffic data, navigation data, etc.). Refer to Chapter II of EOFDM document ‘Breaking the silos’.

### How can the data be visualized?

This can be accomplished with various business intelligence (BI) tools. Content of BI reports vary according to the level of users. These tools usually allow to define data access rules, so that each user will be able to access the data allowed for their account or profile.

The data can be presented in various ways, depending on the available tools:

- ‘Static’ reports, transposing the data for the end user;
- Interactive reports (or dashboards), which will provide some degree of data mining to the end user.

Another important consideration for any report is the amount of data available, particularly digital values. Information overload may have a counter-productive effect on the end users and undermine their ability to interpret the information. Consider showing only the most important information that the report is intended to deliver. For interactive reports or dashboards, ‘mouse over’ features may be a good solution to display further data without adding too much complexity.

## Examples of visual presentation

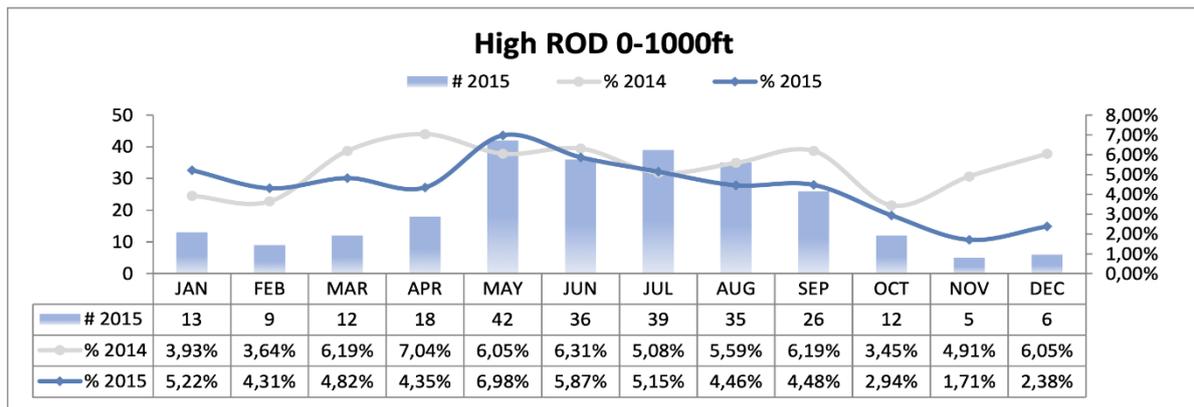
### Bar charts and line charts

Bar charts can be useful for time series. However, for multiple time series, line charts are easier to interpret.

#### Example 1:

Figure 25 summarises statistics of the FDM event ‘high rate of descent (ROD) below 1 000 ft AAL’ by month. The histogram bars represent the absolute number of FDM events (left vertical scale). The blue curve depicts the FDM event rate expressed in percentage for the current year, while the grey line shows the FDM event rate for the previous year (right vertical scale). This example shows the complete picture at the end of the year (December). In practice, the blue line would be continued every month and the graph would only be complete in December. Another possibility is that the time axis is such that the last 12 months are shown, and the last 2 years of operation appear. Figure 25 enables comparing the current situation with the situation at the same period of the previous year. This is a relative assessment. If there has been no major change in the nature of the operation over a year, it signals a trend.

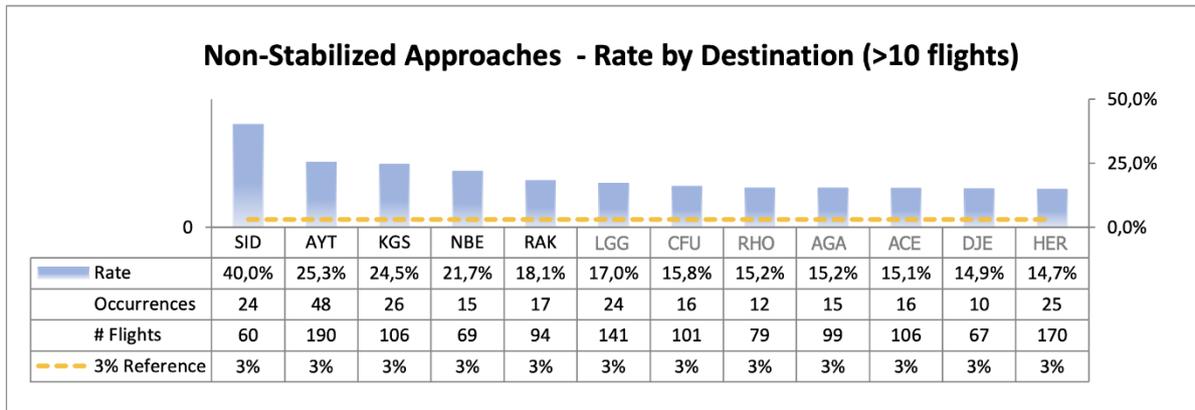
Figure 25 — a combined graph showing occurrences, and year-to-year comparison of event rate (high rate of descent)



#### Example 2:

Figure 26 shows the FDM event rate, number of FDM events, number of flights with a non-stabilised approach by destination (top 12), computed over a year. Additionally, a yellow dashed target line is used as an overlay to indicate the operator’s target for non-stabilised approach. This type of graph helps to identify the destinations with the greatest number of deviations from stabilised approach criteria and may be combined with the colour coding (based on severity) as shown in the following graph.

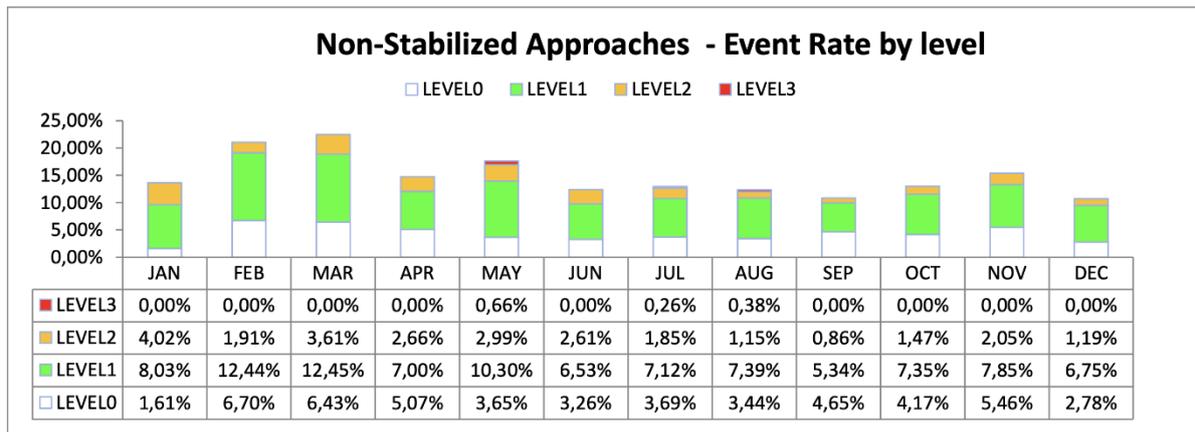
Figure 26 — a combined graph showing event rates by destination and event target



**Example 3:**

Since an FDM event may correspond to a more severe or less severe event, the severity level of each FDM event should be taken into account when monitoring trends over time. Figure 27 shows an example of how these two dimensions might be represented on stacked bar charts (often designated as ‘stacked graphs’). This figure shows the rate of non-stabilised approaches expressed in percentage (or per 1 000) by month. The total height of a block represents the total rate, while the different colours show the severity level. For instance, assuming that the stabilisation height specified in the standard operating procedures is 1 000 ft, white colour is used for LEVEL 0 events (stable > 900 ft), green colour for LEVEL 1 (stable > 750 ft), orange colour for LEVEL 2 (stable > 500 ft) and red colour for LEVEL 3 (stable < 500 ft or never stabilised).

Figure 27 — a stacked graph showing the evolution over time of FDM event rates, by severity level

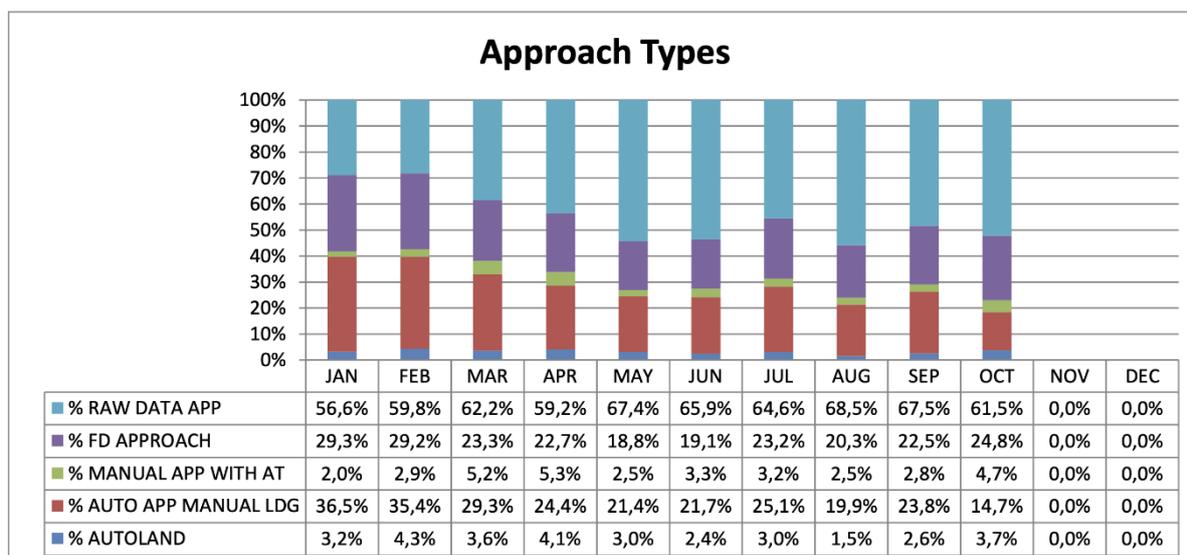


**Example 4**

Figure 28 shows the use of automation expressed as a percentage of all flights by month. The light blue series depicts fully manual approaches (autopilot, auto-thrust, and flight director off), the purple dataset shows flight-director only approaches (autopilot and auto-thrust off), the green series shows manual approaches with auto-thrust on (autopilot off, flight directors off), the red series shows automatic approaches with manual landing (autopilot, auto-thrust, and flight directors on,

disconnected below 1 000 ft) and the dark blue series shows approaches with the autoland system engaged.

Figure 28 — a stacked graph showing category distribution (here approach type) over time



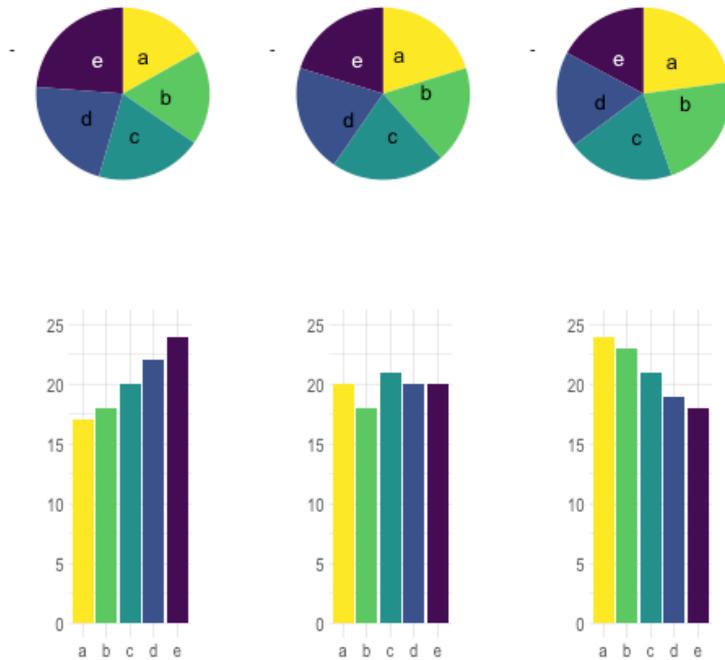
### Pie charts

Pie charts are a popular way to show proportions of a whole and they can be very compelling. However, humans are not good at judging angles (and the associated change in area), therefore representing large amounts of similar data in this way is not helpful. Figure 29 gives an example of how difficult it can be to interpret pie-chart data compared with the same data presented in a bar chart.

There are many alternatives to pie charts including bar charts and pareto charts. However, when using pie charts, a few simple steps can help to keep them informative:

- Make sure that not too many categories are represented (4 or 5 at most);
- Turning the pie chart into a ‘doughnut’ (by adding a hole in the middle) can make it easier to interpret the data as it is more like a length than an area;
- Add clear data labels to inform the proportions and make sure they sum to 100 %; and
- Refrain from making the pie chart three-dimensional, or ‘tilting’ them or ‘exploding’ the segments.

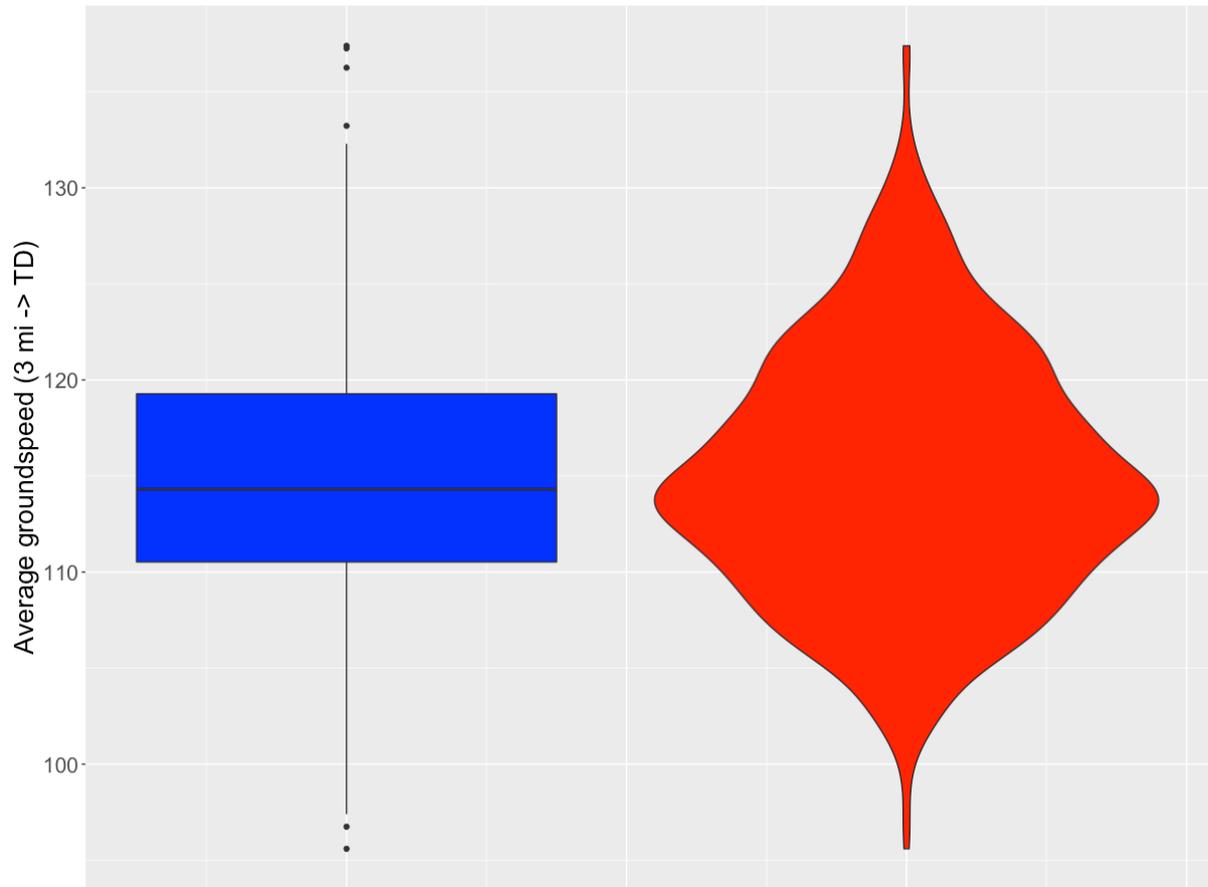
Figure 29 — three distributions represented with pie charts and with bar charts



### Violin Plots

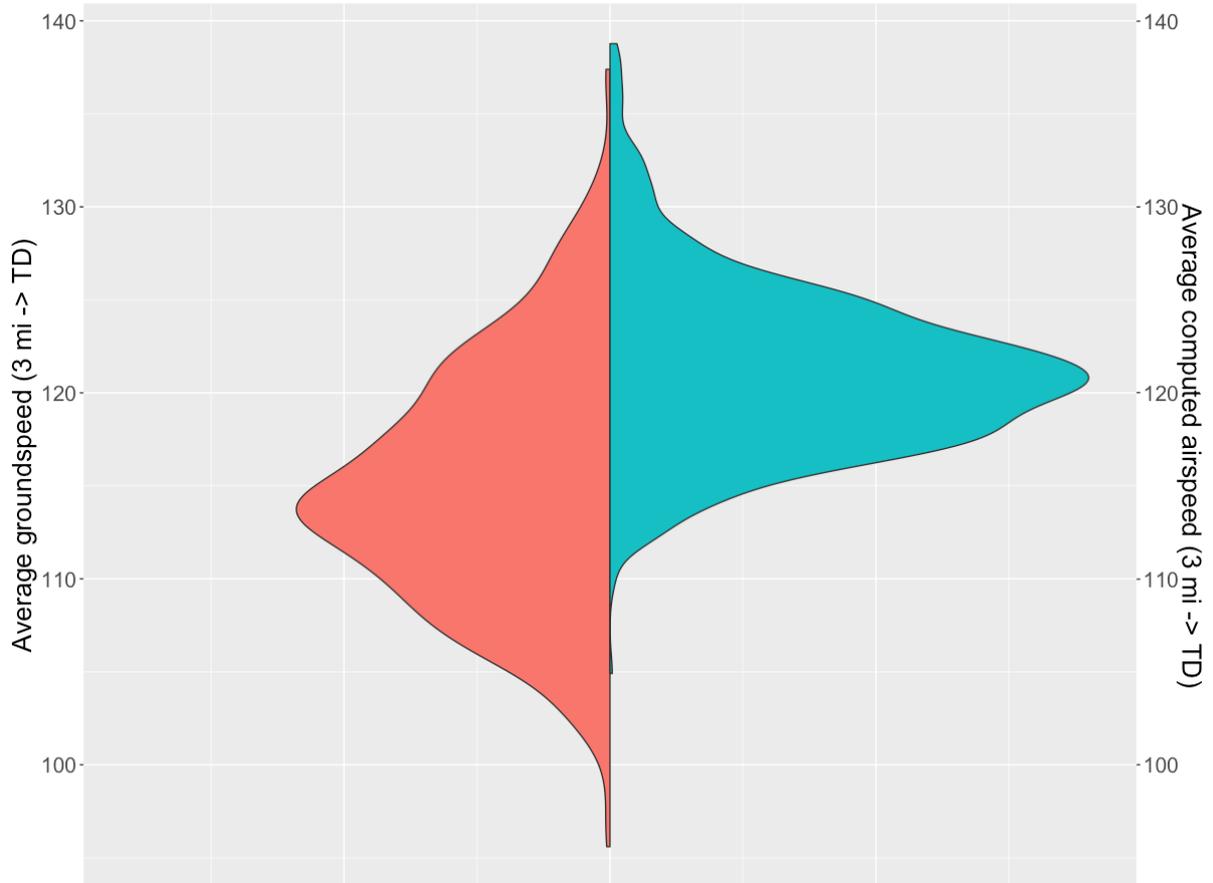
Violin plots are an extension of box plots but represent a continuous distribution rather than selected metrics (e.g. quartiles). In a traditional violin plot, the distribution is presented vertically and mirrored about the vertical axis as shown in Figure 30.

Figure 30 — a boxplot and a violin plot from the same distribution



As an alternative, it is possible to display separate distributions on the left and right, giving a direct comparison, as shown in Figure 31. This is particularly useful where two comparable datasets are analysed.

Figure 31 — violin plots of the average groundspeed and of the average airspeed over the segment from 3 nm to the touchdown point (TD).



*Other examples of graphical presentation*

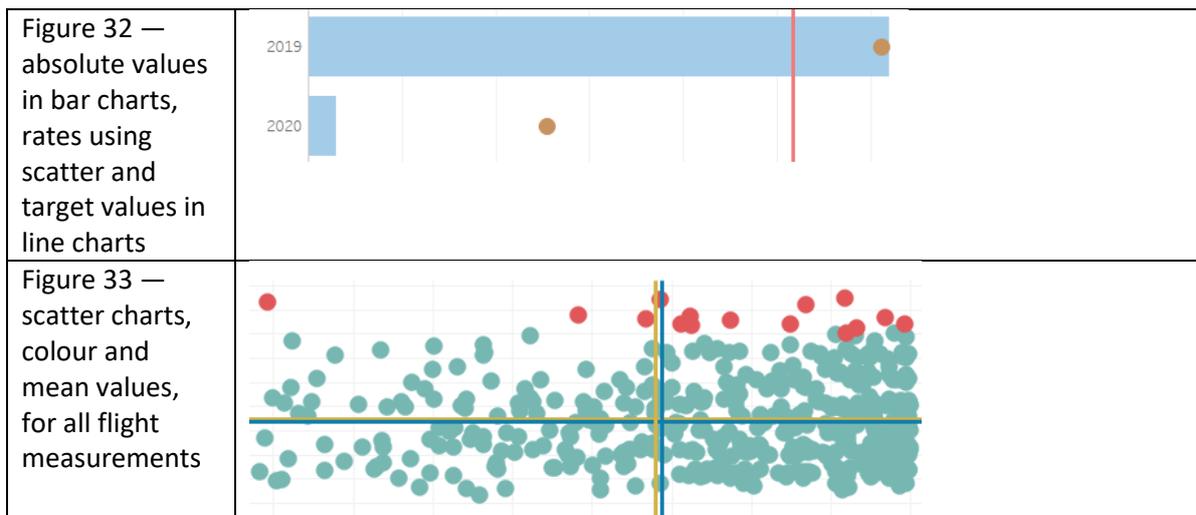


Figure 34 — radar chart  
Example:  
Analysis of unstabilized approaches where the contribution of each triggering condition (configuration, speed, approach path deviation, etc.) can be shown

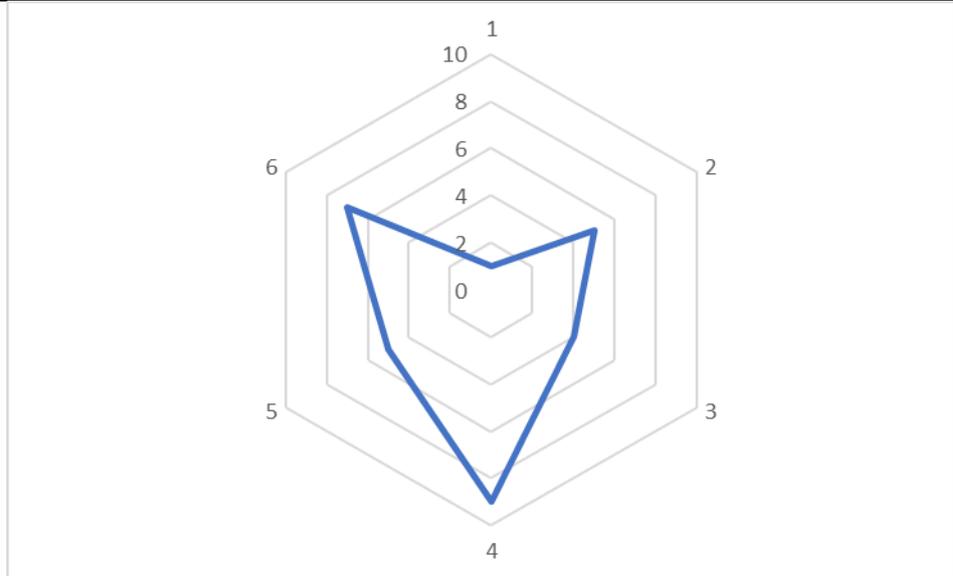


Figure 35 — event allocation by airport, colour, and size can be used to express different characteristics such as rate, absolute values, event severity, etc.



Figure 36 — trajectories extracted from FDM data and shown in a map or a chart

