

FAQs:

[Supplemental Type Certificates \(STC\)](#) , [Applications for product certification/validation of foreign certificates](#), [Certification of products and organisations](#)

Question:

Am I eligible to apply for an STC?

Answer:

Major changes to type design by applicants other than the TC holder must be approved in accordance with Part 21, Section A, Subpart E of Commission Regulation (EC) No 748/2012 i.e. through a Supplemental Type Certificate. Information on the application process can be found at the following [link](#).

Only the STC holder is eligible to apply for a Major Change to STC. For further information, please consult our [website](#). Typically, EASA will issue a revision of the STC.

Minor Changes to an STC may also be applied for by an applicant other than the STC holder. In this case EASA will not issue a revision to the STC but a Minor Change approval in accordance with Subpart D of Commission Regulation (EC) No 748/2012. Information on the application process can be found [here](#).

The following table describes the available options for specific design projects including STCs.

| Type of design | Demonstration of capability | | | |
|--|-----------------------------|--------|-----|------|
| | DOA | ADOA | CP | None |
| Aircraft Type Design | | | | |
| All Aircraft | yes | | | |
| ELA 2* | yes | yes | | |
| ELA 1* | yes | yes | yes | |
| Engine Type Certificate | | | | |
| All Engines | yes | | | |
| Piston Engine | yes | yes | | |
| Engine installed in ELA2 Aircraft | yes | yes | | |
| Engine installed in ELA1 Aircraft | yes | yes | yes | |
| Propeller Type Certificate | | | | |
| All propellers | yes | | | |
| Fixed or adjustable pitch propeller | yes | yes | | |
| Propeller installed in ELA2 Aircraft | yes | yes | | |
| Propeller installed in ELA1 Aircraft | yes | yes | yes | |
| Supplemental Type Certificate (STC) | | | | |
| All STCs | yes | | | |
| STC Group 1** | yes | | | |
| STC Group 2** | yes | yes | | |
| STC on ELA1 or its engine or propeller | yes | yes | yes | |
| Minor Changes | yes | yes | yes | yes |
| Repairs | | | | |
| Minor | yes | yes | yes | yes |
| Major | yes | yes*** | | |
| Major on ELA1 or its engine or propeller | yes | yes | yes | |
| ETSO Authorisation (ETSOA) | yes | yes | | |
| <p>* For definition see EU Regulation (EC) 748/2012 Article 1</p> <p>** For definition see GM 21.A.112B</p> <p>*** Upon Agency agreement</p> | | | | |

Non-EASA Member State applicants for whom a bilateral agreement is in place, no further demonstration of eligibility is required.

Non-DOA/APDOA holders may contact a DOA/ APDOA of their choice to apply for an STC on their behalf.

Design Organisation Approval (DOA): [The process to obtain a DOA](#). Further [information on DOA](#).

Alternative Procedures to Design Organisation Approval (APDOA): [Information on APDOA](#).

However, Part 21.A.14(c) provides the possibility for any natural person to apply for an STC on an ELA 1 aircraft by demonstrating capability through a certification programme. Alternative procedures are not necessary. ELA 1 is generally defined as aircraft with a max MTOW of 1200kg or less, including balloons up to 3400m³ and sailplanes.

| | |
|---|---|
| ELA1 ELA1 aircraft' means the following manned European Light Aircraft: | ELA2 ELA2 aircraft' means the following manned European Light Aircraft: |
|---|---|

| | |
|--|---|
| an aeroplane with a Maximum Take-off Mass (MTOM) of 1 200 kg or less that is not classified as complex motor-powered aircraft | an aeroplane with a Maximum Take-off Mass (MTOM) of 2 000 kg or less that is not classified as complex motor-powered aircraft |
| a sailplane or powered sailplane of 1 200 kg MTOM or less | a sailplane or powered sailplane of 2 000 kg MTOM or less |
| a balloon with a maximum design lifting gas or hot air volume of not more than 3 400 m ³ for hot air balloons, 1 050 m ³ for gas balloons, 300 m ³ for tethered gas balloons | a balloon |
| an airship designed for not more than 4 occupants and a maximum design lifting gas or hot air volume of not more than 3 400 m ³ for hot air airships and 1 000 m ³ for gas airships ⁶ | a hot air airship |
| | a gas airship complying with all of the following characteristics: <ul style="list-style-type: none"> - 3% maximum static heaviness - Non-vector thrust (except reverse thrust) - Conventional and simple design of: structure, control system and ballonnet system - Non-power assisted controls |
| | a Very Light Rotorcraft |

| | |
|--|--|
| Certification Programme Demonstration of capability via a certification programme for: | AP DOA Demonstration of capability via AP DOA for: |
| ELA1 aircraft | ELA2 aircraft |
| Engine [to be] installed in ELA1 aircraft | Engine [to be] installed in ELA2 aircraft |

| | |
|--|--|
| Propeller [to be] installed in ELA1 aircraft | Propeller [to be] installed in ELA2 aircraft |
| | Piston Engine |
| | Fixed or adjustable pitch propeller |

Last updated:

17/03/2020

Link:<https://www.easa.europa.eu/lv/faq/48612>