



**European Union Aviation Safety Agency**

---

**EASA**

**TYPE-CERTIFICATE  
DATA SHEET**

**EASA.A.024**

**L-13 "BLANÍK"**

Type Certificate Holder:

**Blanik Aircraft CZ s.r.o.**  
Beranových 65  
199 00 Praha 9 - Letňany  
Czech Republic

For models: L-13 "BLANÍK"  
L - 13 AC BLANÍK  
L 13 A Blaník

Issue 11: 05 December 2022

## **0.I. Table of Content**

### **SECTION 0:**

- 0.I. Table of Content

### **SECTION A: L-13 "BLANÍK"**

- A.I. General
- A.II. Certification Basis
- A.III. Technical Characteristics and Operational Limitations
- A.IV. Operating and Service Instructions
- A.V. Notes

### **SECTION B: L - 13 AC BLANÍK**

- B.I. General
- B.II. Certification Basis
- B.III. Technical Characteristics and Operational Limitations
- B.IV. Operating and Service Instructions
- B.V. Notes

### **SECTION C: L 13 A Blaník**

- C.I. General
- C.II. Certification Basis
- C.III. Technical Characteristics and Operational Limitations
- C.IV. Operating and Service Instructions
- C.V. Notes

### **ADMINISTRATIVE SECTION**

- I. Acronyms
- II. Type Certificate Holder Record
- III. Change Record

**A.I. General (L - 13 "BLANÍK")**

- |  |  |                        |
|--|--|------------------------|
| Data Sheet No.: EASA.A.024   | Issue: 11  | Date: 05 December 2022 |
| 1. Model:  | L - 13 "BLANÍK"  |                        |
| 2. Airworthiness Category:   | Acrobatic<br>Cloud flying  |                        |
| 3. Type Certificate Holder:  | Blanik Aircraft CZ s.r.o.<br>Beranových 65<br>190 00 Praha 9 - Letňany<br>Czech Republic   |                        |
| 4. Manufacturer:   | Strojírny první pětiletky n.p.<br>Uherské Hradiště - Kunovice 1177<br>Czechoslovakia:<br><br>from S/N 170101 to S/N 170920,<br>from S/N 171001 to S/N 171930,<br>from S/N 172001 to S/N 172960,<br>from S/N 173001 to S/N 173930,<br>from S/N 174001 to S/N 174930,<br>from S/N 175001 to S/N 175330,<br><br>LET, n.p.<br>Uherské Hradiště - Kunovice 1177<br>Czechoslovakia<br><br>from S/N 025401 to S/N 025930,<br>from S/N 026001 to S/N 026960,<br>from S/N 027001 to S/N 027361,<br>from S/N 817401 to S/N 817403,<br>from S/N 827404 to S/N 827421, |                        |
| 5. Certification Application Date:                                       | ---  |                        |
| 6. CAA CZ Type Certification Date:                                       | May 29, 1959   |                        |
| 7. EASA Type Certification Date:   | February 4, 2005   |                        |
| 8. The EASA Type Certificate replaces Czech Type Certificate No. 2725-59 |  |                        |

## **A.II. Certification Basis**

1. Reference Date for determining the applicable requirements: ---
2. Certification Basis: - Bauvorschriften für Segelflugzeuge (BSV) issued August 1939  
- BCAR, Section E, issued June 16, 1966
3. Airworthiness Requirements: see (2) above
4. Requirements elected to comply: CS 22.627, Amdt. 2 (Fatigue strength) for sailplanes with installed modification as per Mandatory Bulletin L13/117a initial issue (covering Major Change TDC-002-L13-BL) or its further approved revisions.
5. EASA Special Conditions: None
6. EASA Exemptions: None
7. EASA Equivalent Safety Findings: None

## **A.III. Technical Characteristics and Operational Limitations**

1. Type Design Definition: Drawing No. A 101 310 N
2. Description: All-metal, cantilever, high-wing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake, and a tail skid or (an optional) tail wheel. Horizontal tail surfaces consist of a two-piece tailplane and elevator, vertical tail surfaces consist of a fin and a rudder.
3. Equipment: Minimum equipment:  
2 airspeed indicators, range to 400 km/h  
2 altimeters  
2 four-point safety harnesses (symmetric)  
2 parachutes or backrests (approx. 10 cm thick when compressed)
4. Dimensions:

Span	16.2 m
Length	8.4 m
Height	2.1 m
Wing Area	19.15 m <sup>2</sup>
Aspect Ratio	13,7
5. Launching Hooks: Nose tow release Dwg. No. A 740 210 N, or nose tow release "E85", LBA Type Certificate No. 60.230/1  
Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P  
"Europa G 72" safety tow release LBA Type Certificate No. 60.230/2, or  
"Europa G 73" safety tow release LBA Type Certificate No. 60.230/2, or  
"Europa G 88" safety tow release LBA Type Certificate No. 60.230/2

- |  |   |                            |
|--|---|----------------------------|
| 6. Weak links:                             | Ultimate Strength:<br>- for winch launching<br>- for aero-tow                                 | max. 6300 N<br>max. 6300 N |
| 7. Air Speeds:                             |   |                            |
| Manoeuvring Speed $V_A$                    | 145 km/h IAS  |                            |
| Never Exceed Speed $V_{NE}$                |   |                            |
| up to 2500 m MSL                           | 253 km/h IAS  |                            |
| 2500 to 3000 m MSL                         | 245 km/h IAS  |                            |
| 3000 to 4000 m MSL                         | 230 km/h IAS  |                            |
| 4000 to 5000 m MSL                         | 215 km/h IAS  |                            |
| 5000 to 6000 m MSL                         | 201 km/h IAS  |                            |
| 6000 to 7000 m MSL                         | 187 km/h IAS  |                            |
| 7000 to 8000 m MSL                         | 174 km/h IAS  |                            |
| 8000 to 9000 m MSL                         | 161 km/h IAS  |                            |
| 9000 to 10000 m MSL                        | 150 km/h IAS  |                            |
| Rough Air Speed $V_{RA}$                   | 145 km/h IAS  |                            |
| Max. Aerotow Speed $V_T$                   | 140 km/h IAS  |                            |
| Max. Flap Extended Speed $V_{FE}$          | 110 km/h IAS  |                            |
| Max. Winch-launch Speed $V_W$              | 120 km/h IAS  |                            |
| Max. Landing Gear Operating Speed $V_{LO}$ | ---   |                            |
| 8. Operational Capability:                 | VFR Day<br>Cloud flying   |                            |
| 9. Maximum Weights:                        |   |                            |
| Maximum Take-Off Weight:                   | 500 kg  |                            |
| Maximum Weight of non-lifting parts:       | 355 kg  |                            |
| Empty Weight:                              | 292 kg $\pm$ 2% original unreinforced design<br>306 kg $\pm$ 2% reinforced as per MB L13/117a |                            |
| 10. Centre of Gravity Range:               |   |                            |
|  | Fore most c.g. limit aft of reference plane 112 mm  |                            |
|  | Aft most c.g. limit aft of reference plane 300 mm   |                            |
|  | [MAC is 1253 mm]  |                            |
| 11. Datum:                                 | Wing leading edge at root rib   |                            |
| 12. Levelling Means:                       | Leveling points on fuselage   |                            |
| 13. Minimum Flight Crew:                   | 1 (Pilot)   |                            |
| 14. Maximum Passenger Seating Capacity:    | 1   |                            |
| 15. Lifetime limitations:                  | Refer to Maintenance Manual   |                            |
| 16. Deflection angles of control surfaces: |   |                            |
| Elevator                                   | up  | 32° + 2°                   |
|  | down  | 25° $\pm$ 1°               |
| Rudder                                     | left, right   | 30° + 1°                   |
| Ailerons                                   | up  | 34° + 2°                   |
|  | down  | 13° + 2°                   |
| Wing flaps                                 | down  | 08° $\pm$ 1°               |
| Elevator trim tab                          | up  | 12° $\pm$ 1°               |
|  | down  | 35° $\pm$ 1°               |

## **A.IV. Operating and Service Instructions**

### 1. Flight Manual:

In Czech language Do-L13-1111.1	Směrnice pro pilota větroně L 13
In English language Do-L13-1111.3	Pilots Notes for the L-13 Sailplane
In German language Do-L13-1111.2	Fluganweisung für das Segelflugzeug L-13
In Russian language Do-L13-1111.5	Rukovodstvo po letnoj ekspluatácii planera L 13
In Spanish language Do-L13-1111.4	Planeador L 13 Blaník Instruciones Para el Piloto
In English language Do-L13-1111.6	L-13 „BLANÍK“ Sailplane Flight Manual *)
*)	For sailplanes with installed modification as per Mandatory Bulletin L13/117a initial issue (covering Major Change TDC-002-L13-BL) or its further approved revisions.

### 2. Maintenance Manual:

In Czech language Do-L13-1132.1 Do-L13-1131.1	Technická příručka větroně L 13 Příručka pro provoz a údržbu větroně L 13 Blaník bez generálních oprav
In English language Do-L13-1132.3 Do-L13-1131.3	Technical Manual of the L 13 Sailplane, Manual for Operation and Maintenance of the L 13 Blaník Sailplane without overhauls
In German language Do-L13-1132.2	Technisches Handbuch für das Segelflugzeug L-13
In Russian language Do-L13-1131.5	Rukovodstvo po techničeskoj ekspluatácii L 13 Blaník bez kapitalnych remontov

### 3. Illustrated Parts Catalogue:

In Czech language Do L13-2121.6	Kusovník větroně L 13 Blaník (C-A-N)
In English language Do-L13-2121.6	Spare Parts Catalogue L 13 Blaník (C-A-N)
In German language Do-L13 2121.6	Katalog der Bestandteile L-13 Blaník

### 4. Overhaul Manual

In Czech language Do-L13-3031.1	Příručka pro generální opravu kluzáku L 13, L 13A
In English language Do-L13-3031.3	Overhaul Manual for L 13, L 13A Gliders
In Russian language Do-L13-3131.5	Rukovodstvo po kapitalnomu remontu planera L 13

### 5. Operating Manuals for Tow Releases

In German and in English language	Operating Manual for Nose Tow Releases TOST "Europa E 85"
	Operating Manual for Safety Tow Releases TOST "Europa G 88"
	Operating Manual for Safety Tow Releases TOST "Europa G 72" and "Europa G 73"

### **A.V. Notes**

1. Six-position Serial Numbers starting with 17 precede the six-position Serial Numbers starting with 02.
2. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.
3. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.
4. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.

**B.I. General (L - 13 AC BLANÍK)**

Data Sheet No.: EASA.A.024	Issue: 11	Date: 05 December 2022
1. Model:	L - 13 AC BLANÍK	
2. Airworthiness Category:	Acrobatic Cloud flying	
3. Type Certificate Holder:	Blanik Aircraft CZ s.r.o. Beranových 65 190 00 Praha 9 - Letňany Czech Republic	
4. Manufacturer:	LET, a.s. 686 04 Kunovice 1177 CZECH REPUBLIC  from S/N 988601 to S/N 988604 from S/N 008605 to S/N 008606  LETECKÉ ZÁVODY a.s. 686 04 Kunovice 1177 CZECH REPUBLIC  S/N 018901 and from S/N 028902 to S/N 028905 and S/N 029101	
5. Certification Application Date:	---	
6. CAA CZ Type Certification Date:	July 15, 1999	
7. EASA Type Certification Date:	February 4, 2005	
8. The EASA Type Certificate replaces Czech Type Certificate No. 2725-59		

**B.II. Certification Basis**

1. Reference Date for determining the applicable requirements:	August 31, 1998
2. Certification Basis:	CRI-A-01, issue 2, issued August 31, 1998
3. Airworthiness Requirements:	BCAR, Section E, issued June 6, 1966
4. Requirements elected to comply:	None
5. EASA Special Conditions:	Appendix H, Joint Airworthiness Requirements, Sailplanes and Powered Sailplanes, Change 5 of October 28, 1995
6. EASA Exemptions:	None
7. EASA Equivalent Safety Findings:	None



### **B.III. Technical Characteristics and Operational Limitations**

1. Type Design Definition:	Drawing No. A 500 020 N
2. Description:	L - 13 AC BLANÍK sailplane is all-metal, cantilever, highwing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake and a tail wheel. Horizontal tail surfaces consist of a two-piece tailplane and elevator, vertical tail surfaces consist of a fin and a rudder. The S/N 018901 and from S/N 039102 there are installed ailerons outer stops, left aileron balance tab and there exists an option of wing tip extensions installation. Among the standard equipment there belongs AMU-1B recording unit.
3. Equipment:	Minimum equipment: 2 airspeed indicators, range to 400 km/h 2 altimeters 2 five-point safety harnesses (symmetric) 2 accelerometers 2 parachutes or backrests (approx. 10 cm thick when compressed) 1 AMU-1B recording unit
4. Dimensions:	
Span	14.2 m without wing-tip extensions 16.2 m with wing-tip extensions
Length	8.4 m
Height	2.09 m
Wing Area	17.44 m <sup>2</sup> without wing-tip extensions 19.15 m <sup>2</sup> with wing-tip extensions
Aspect Ratio	11.186 without wing-tip extensions 13.7 with wing-tip extensions
5. Launching Hooks:	Nose tow release Dwg. No. A 740 210 N, or nose tow release "E85", LBA Type Certificate No. 60.230/1 Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P "Europa G 88" safety tow release LBA Type Certificate No. 60.230/2
6. Weak links:	Ultimate Strength: - for winch launching                      max. 6230 N - for aero-tow                               max. 6230 N
7. Air Speeds:	Air speeds of sailplane without wing tip extensions:
Manoeuvring Speed $V_A$	160 km/h IAS
Never Exceed Speed $V_{NE}$	
up to 2500 m MSL	230 km/h IAS
2500 to 3000 m MSL	223 km/h IAS
3000 to 4000 m MSL	209 km/h IAS
4000 to 5000 m MSL	195 km/h IAS
5000 to 6000 m MSL	182 km/h IAS
6000 to 7000 m MSL	170 km/h IAS
7000 to 8000 m MSL	158 km/h IAS
8000 to 9000 m MSL	147 km/h IAS

9000 to 10000 m MSL	136 km/h IAS
Rough Air Speed $V_{RA}$	160 km/h
Max. Aerotow Speed $V_T$	150 km/h
Max. Winch-launch Speed $V_W$	120 km/h
Max. Landing Gear Operating Speed $V_{LO}$	---

	Air speeds of sailplane with wing tip extensions:
Maneuvering Speed $V_A$	150 km/h IAS
Never Exceed Speed $V_{NE}$	
up to 2500 m MSL	230 km/h IAS
2500 to 3000 m MSL	223 km/h IAS
3000 to 4000 m MSL	209 km/h IAS
4000 to 5000 m MSL	195 km/h IAS
5000 to 6000 m MSL	182 km/h IAS
6000 to 7000 m MSL	170 km/h IAS
7000 to 8000 m MSL	158 km/h IAS
8000 to 9000 m MSL	147 km/h IAS
9000 to 10000 m MSL	136 km/h IAS
Rough Air Speed $V_{RA}$	150 km/h
Max. Aerotow Speed $V_T$	150 km/h
Max. Winch-launch Speed $V_W$	120 km/h
Max. Landing Gear Operating Speed $V_{LO}$	---

8. Operational Capability: VFR Day  
Cloud flying
9. Maximum Weights:
- |                                      |                                      |
|--------------------------------------|--------------------------------------|
| Maximum Weight:                      | 500 kg (without extensions)          |
|                                      | 510 kg (with extensions)             |
| Maximum Weight of non-lifting parts: | 355 kg                               |
| Empty Weight:                        | 305 kg $\pm$ 2% (without extensions) |
|                                      | 315 kg $\pm$ 2% (with extensions)    |
10. Centre of Gravity Range: Fore most c.g. limit aft of reference plane 143 mm  
Aft most c.g. limit aft of reference plane 337 mm  
[MAC is 1295 mm without wing tip extensions]  
[MAC is 1252.3 mm with wing tip extensions]
11. Datum: Wing leading edge at root rib
12. Levelling Means: Leveling points on fuselage in horizontal position.
13. Minimum Flight Crew: 1 (Pilot)
14. Maximum Passenger Seating Capacity: 1
15. Lifetime limitations: Refer to Maintenance Manual
16. Deflection angles of control surfaces:
- |                          |             |              |
|--------------------------|-------------|--------------|
| Elevator                 | up          | 32° + 2°     |
|                          | down        | 27° $\pm$ 1° |
| Rudder                   | left, right | 29° + 1°     |
| Ailerons                 | up          | 34° + 2°     |
|                          | down        | 13° + 2°     |
| Left aileron balance tab | up          | 20° $\pm$ 2° |
|                          | down        | 15° $\pm$ 2° |
| Elevator trim tab        | up          | 12° $\pm$ 1° |
|                          | down        | 35° $\pm$ 1° |

## **B.IV. Operating and Service Instructions**

### 1. Flight Manual:

In Czech language Do-L13AC-1013.1	Letová příručka L 13 AC Blaník (to S/N 008606 from S/N 028902 to 029101)
Do-L13AC-1014.0	Letová příručka L 13 AC Blaník (S/N 018901 from S/N 039102)
In English language Do-L13AC-1013.3	Sailplane Flight Manual L 13AC Blaník to S/N 008606 from S/N 028902 to 029101
Do-L13AC-1014.2	Sailplane Flight Manual L 13AC Blaník (S/N 018901 from S/N039102)

### 2. Maintenance Manual:

In Czech language Do-L13AC-1032.1	Provozně technická příručka kluzáku L 13 AC
In English language Do-L13AC-1032.3	Maintenance Manual for the Sailplane L 13 AC

### 3. Illustrated Parts Catalogue:

In Czech language Do-L13AC-2051.0	Katalog dílů a montážních jednotek kluzáku L-13AC Blaník (C/A)
In English language Do-L13AC-2051.0	Illustrated parts catalogue for the sailplane L 13 AC Blaník (C/A)

### 4. Operation book

In Czech language Do-L13AC.1015.02	L 13AC Blaník Záznamník provozu kluzáku (C/A)
In English language Do-L13AC.1015.02	L 13 AC Blaník Sailplane operation book of records (C/A)

### 5. Operating Manuals for Tow Releases

In German and in English language	Operating Manual for Nose Tow Releases TOST "Europa E 85"
	Operating Manual for Safety Tow Releases TOST "Europa G 88"

## **B.V. Notes**

1. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.
2. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.
3. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.

**C.I. General (L 13 A Blaník)**

Data Sheet No.: EASA.A.024	Issue: 11	Date: 05 December 2022
1. Model:	L 13 A Blaník	
2. Airworthiness Category:	Acrobatic Cloud flying	
3. Type Certificate Holder:	Blaník Aircraft CZ s.r.o. Beranových 65 190 00 Praha 9 - Letňany Czech Republic	
4. Manufacturer:	LET, n.p. 686 04 Kunovice 1177 CZECH REPUBLIC  from S/N 817401 to S/N 827421  LET, a.s. 686 04 Kunovice 1177 CZECH REPUBLIC  from S/N 968501 to S/N 968505	
5. Certification Application Date:	---	
6. CAA CZ Type Certification Date:	December 16, 1981	
7. EASA Type Certification Date:	12 August 2005	
8. The EASA Type Certificate replaces Czech Type Certificate No. 2725-59		

**C.II. Certification Basis**

1. Reference Date for determining the applicable requirements:	---
2. Certification Basis:	BCAR, Section E, issued June 6, 1966
3. Airworthiness Requirements:	see (2) above
4. Requirements elected to comply:	None
5. EASA Special Conditions:	None
6. EASA Exemptions:	None
7. EASA Equivalent Safety Findings:	None

### **C.III. Technical Characteristics and Operational Limitations**

1. Type Design Definition:	Drawing No. A 101 310 N
2. Description:	All-metal, cantilever, high-wing monoplane. Wing fitted with wing flaps and DFS air brakes. Landing gear consists of a semi-retractable landing wheel with a mechanical brake, and a tail skid or (an optional) tail wheel. Horizontal tail surfaces consist of a two-piece tailplane and elevator, vertical tail surfaces consist of a fin and a rudder.
3. Equipment:	Minimum equipment: 2 airspeed indicators, range to 400 km/h 2 altimeters 2 four-point safety harnesses (symmetric) 2 parachutes or backrests (approx. 10 cm thick when compressed)
4. Dimensions:	
Span	16.2 m
Length	8.4 m
Height	2.1 m
Wing Area	19.15 m <sup>2</sup>
Aspect Ratio	13.7
5. Launching Hooks:	Nose tow release of type SR-L13.225, or nose tow release Dwg. No. A 740 210 N, or nose tow release "E85", LBA Type Certificate No. 60.230/1 Side tow release left Dwg. No. LN-0399L and right Dwg. No. LN-0400P "Europa G 72" safety tow release LBA Type Certificate No. 60.230/2, or "Europa G 73" safety tow release LBA Type Certificate No. 60.230/2, or "Europa G 88" safety tow release LBA Type Certificate No. 60.230/2
6. Weak links:	Ultimate Strength: - for winch launching      max. 6230 N - for aero-tow              max. 6230 N
7. Air Speeds:	
Manoeuvring Speed $V_A$	145 km/h IAS
Never Exceed Speed $V_{NE}$	
up to 2500 m MSL	253 km/h IAS
2500 to 3000 m MSL	245 km/h IAS
3000 to 4000 m MSL	230 km/h IAS
4000 to 5000 m MSL	215 km/h IAS
5000 to 6000 m MSL	201 km/h IAS
6000 to 7000 m MSL	187 km/h IAS
7000 to 8000 m MSL	174 km/h IAS
8000 to 9000 m MSL	161 km/h IAS
9000 to 10000 m MSL	150 km/h IAS
Rough Air Speed $V_{RA}$	145 km/h
Max. Aerotow Speed $V_T$	140 km/h
Max. Winch-launch Speed $V_W$	120 km/h
Max. Landing Gear Operating Speed $V_{LO}$	---

8. Operational Capability:	VFR Day Cloud flying		
9. Maximum Weights:			
Maximum Weight:	500 kg		
Maximum Weight of non-lifting parts:	355 kg		
Empty Weight:	306 kg ± 2%		
10. Centre of Gravity Range:	Fore most c.g. limit aft of reference plane 112 mm Aft most c.g. limit aft of reference plane 300 mm [MAC is 1253 mm]		
11. Datum:	Wing leading edge at root rib		
12. Levelling Means:	Leveling points on fuselage in horizontal position.		
13. Minimum Flight Crew:	1 (Pilot)		
14. Maximum Passenger Seating Capacity:	1		
15. Lifetime limitations:	Refer to Maintenance Manual		
16. Deflection angles of control surfaces:			
	Elevator	up	32° + 2°
		down	25° ± 1°
	Rudder	left, right	30° + 1°
	Ailerons	up	34° + 2°
		down	13° + 2°
	Wing flaps	down	8° ± 1°
	Elevator trim tab	up	12° ± 1°
		down	35° ± 1°

## **C.IV. Operating and Service Instructions**

### 1. Flight Manual:

In Czech language Do-L13A-1011.1	Letová příručka L 13 A
In English language Do-L13A-1011.3	Flight Manual of the L 13 A
In German language Do-L13A-1011.2	Flughandbuch für das Segelflug L 13 A

### 2. Maintenance Manual:

In Czech language Do-L13A-1031.1	Příručka pro obsluhu, údržbu a opravy kluzáku L 13
In English language Do-L13A-1031.3	Technical Manual of the L 13 A Sailplane
In German language Do-L13A-1031.2	Handbuch für die Reparatur, Instandhaltung und Wartung des Segelflugzeuges L 13 A

3. Illustrated Parts Catalogue:

In Czech language Do-L13-2126.6	Kusovník větroně L 13 Blaník (C-A-N)
In English language Do-L13-2126.6	Spare Parts Catalogue L 13 Blaník (C-A-N)
In German language Do-L13-2126.6	Katalog der Bestandteile L-13 Blaník (C-A-N)

4. Overhaul Manual

In Czech language Do-L13-3031.1	Příručka pro generální opravu kluzáku L 13, L 13A
In English language Do-L13-3031.3	Overhaul Manual for L 13, L 13A Gliders

5. Sailplane Technical Description L 13A

In Czech language Do-L13A-1021.1	Sailplane Technical Description L 13A
-------------------------------------	---------------------------------------

6. Operating Manuals for Tow Releases

In German and in English language	Operating Manual for Nose Tow Releases TOST "Europa E 85"
	Operating Manual for Safety Tow Releases TOST "Europa G 88"
	Operating Manual for Safety Tow Releases TOST "Europa G 72" and "Europa G 73"

**C.V. Notes**

1. Since 20 June 2013 the TC holder obligations are covered by an agreement signed between new TC holder (BLANIK LIMITED) and Contracted DOA Holder (Aircraft Industries a.s.). For Continuing Airworthiness and other technical issues contact directly the Contracted DOA Holder.
2. Since 30 September 2016 the TC holder obligations are covered by an agreement signed between TC holder (BLANIK LIMITED) and Contracted DOA Holder (Blanik Aircraft CZ s.r.o. / EASA.21J.609). At the same time a contract between TC holder and Aircraft Industries a.s. / EASA.21J.119) was terminated. For Continuing Airworthiness and other technical issues contact directly the new Contracted DOA Holder.
3. On 10 January 2017, Blanik Aircraft CZ s.r.o. / EASA.21J.609 became the TC holder.

## **ADMINISTRATIVE SECTION**

### I. Acronyms

N/A

### II. Type Certificate Holder Record

#### Up to 19 June 2013

Aircraft Industries, a.s.  
Na Záhonech 1177  
686 04 Kunovice  
Czech Republic

#### Up to 09 January 2017

BLANIK LIMITED  
2<sup>nd</sup> Floor Beaux Lane House  
Mercer Street Lower  
Dublin 2  
Republic of Ireland

#### Up to 12 September 2019

Blanik Aircraft CZ s.r.o.  
Karolinská 661/4  
186 00 Praha 8  
Czech Republic

#### Since 13 September 2019

Blanik Aircraft CZ s.r.o.  
Beranových 65  
190 00 Praha 9 - Letňany  
Czech Republic



### III. Change Record

Issue	Date	Changes
01	04 February 2005	Initial issue of TCDS No. EASA.A.044
02	12 August 2005	To record the change in the name of the TC holder and to include the L 13 A model.
03	09 September 2005	Editorial changes
04	23 May 2006	Change in address of the TC holder
05	20 June 2013	Change of the TC holder and new layout of TCDS
06	09 May 2016	Addition of S/N 827421 (L 13 A Blanik) to section C.I.5.
07	21 September 2016	Addition of CS 22.627 Eltect to Comply to section A.II.4. and addition of Sailplane Flight Manual (Do-L13-1111.6) to section A.IV.1.
08	30 September 2016	Change of the Contracted DOA
09	10 January 2017	Change of the TC holder
10	13 September 2019	Change of the TC holder's address
11	05 December 2022	Correction of S/N per manufacturer and correction of Maximum Empty Weight