

European Aviation Safety Agency

EASA TYPE-CERTIFICATE DATA SHEET

L 200

Type Certificate Holder:

Aircraft Industries, a.s

Na Záhonech 1177
686 04 Kunovice
CZECH REPUBLIC

Manufacturer:

Aircraft Industries, a.s

Na Záhonech 1177
686 04 Kunovice
CZECH REPUBLIC

Strojírny první pětiletky n.p.

686 04 Kunovice
CZECH REPUBLIC

For variants: L – 200 A
L – 200 D

Issue 01: 12 August 2005
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Change record

SECTION A1: GENERAL, L-200 A Type Design

A1. General

Data Sheet No.: EASA.A.043	Issue: 04	Date: 24 May 2006
1. a) Type:	L 200	
b) Variant:	L 200 A	
2. Airworthiness Category:	Normal	
3. Type Certificate Holder::	Aircraft Industries, a.s Na Záhonech 1177 686 04 Kunovice Czech Republic	
4. Manufacturer:	Strojírny první pětiletky n.p. 686 04 Kunovice Czech Republic	
5. Certification Application Date:	---	
6. CAA CZ Type Certification Date::	July 27,1960	
7. EASA Type Certification Date	August 12, 2005 (see Note 1)	
8. The EASA Type Certificate replaces	Czech Type Certificate No.: 6208/60	

A11. Certification Basis

1. Reference Date for determining the the applicable requirements	---
2. Certification Basis:	---
3. Airworthiness Requirements:	British Civil Airworthiness Requirements (BCAR)normal category, Section D, C (Amendment No. 6 289/62) Civil Air Regulations (CAR) Part 3, normal category, Bauvorschriften für Flugzeuge (BVF) usage group P – transport, loading class 3 – normal
4. Requirements elected to comply:	None

- | | |
|-------------------------------------|------|
| 5. EASA Special Conditions: | None |
| 6. EASA Exemptions: | None |
| 7. EASA Equivalent Safety Findings: | None |
| 8. EASA Environmental Standards: | None |

All. Technical Characteristics and Operational Limitations

- | | |
|----------------------------|---|
| 1. Type Design Definition: | The specification list of Aircraft L 200 A
Aircraft Drawing No L200.0000 |
| 2. Description: | Two-engines, five-seater, self-supporting,
low-wing aircraft of all-metal design. |
| 3. Equipment: | Flight and navigation instruments:
Airspeed indicator up to 400 km/hr
Altimeter up to 10 km
Variometer
Turn and bank indicator
Combined artificial horizon
Compass
Radio station
Direction gyro

Engine instruments:
Dual RPM indicator
Dual indicator of filling pressure
Quadruplicate indicator of engine
parameters
Dual indicator of cylinder heads
temperature

Other instruments:
Landing gear position indicator
Flaps position indicator
Fuel quantity in main tank indicator
Indicator of remaining fuel in tanks
Pressure gauge of hydraulic brakes
Triplicate indicator of heating temperature
Volt-ammeter
Clock |
| 4. Dimensions: | |
| | Span 12.30 m |
| | Length 8.60 m |
| | Height 2.215 m |
| | Wing Area 17.28 m ² |

5. Engines:

5.1. Model:	M 337,
5.2. Type Certificate:	94-06, issued by CAA CZ
5.3. Limitations:	
Maximum take-off power:	
Max. Power	155 kW (210 HP)
Max. Engine speed	2750 RPM
Maximum continuous (nominal) power:	
Max. Power	121 kW (165 HP)
Max. Engine speed	2 600 RPM
Maximum cruising power:	
Max. Power	83,7 kW (132,5 HP)
Max. Engine speed	2400 RPM

6. (reserved)

7. Propellers:

7.1.1. Model:	V 410AT
7.1.2. Type Certificate:	9 339/60, issued by CAA CZ
7.1.3. Number of blades:	2
7.1.4. Diameter:	1900 mm
7.1.5. Sense of rotation:	Rotating anti-clock-wise in the view of the flight direction
	or
7.2.6. Model:	V 410 T
7.2.7. Type Certificate:	9 339/60, issued by CAA CZ
7.2.8. Number of blades:	2
7.2.9. Diameter:	1900 mm
7.2.10. Sense of rotation:	Rotating anti-clock-wise in the view of the flight direction

8. Fluids:

8.1. Fuel	ESSO ICP 80 SHELL Avgas 80 SHELL Avgas 100LL BP 100 L BL 78 according to ČSN 65 6510
8.2. Oil	AEROSHELL Oil W 100 AEROSHELL Oil W 120 ELF Aviation AD 100 MOBIL Aero D 100 BP Aero Oil 100 CASTROL Aero AD 100 TOTAL Aero D100

7. Fluid capacities

7.1. Fuel: 230 litres (2x115 litres)
420 litres (with auxiliary fuel tanks 2x95 l)

7.2. Oil: 2x12 litres

8. Air Speeds:

Manoeuvring Speed	V _A	228	km/h	IAS
Never Exceed Speed	V _{NE}	360	km/h	IAS
Normal operating speed	V _{NO}	280	km/h	IAS
Maximum flaps extended speed	V _{FE}	170	km/h	IAS

9. Maximum Operating Altitude: 5700 m

10. Operational Capability: VFR and IFR flights.

11. Weights:

Maximum Weight:	1950	kg
Empty Weight:	1275	kg

12. Centre of Gravity Range: 15,3 % – 29 % M.A.C. (M.A.C = 1497mm)

13. Datum: Passes by the fuselage forward leveling items perpendicularly to the aircraft longitudinal axes

14. (reserved)

15. Levelling Means: Leveling level is identical with the datum - see leveling diagram shown in the L 200 A Aircraft Maintenance Manual.

16. Minimum Flight Crew: 1 (Pilot)

17. Maximum Passenger Seating Capacity: 4

18. (reserved):

19. Baggage / Cargo Compartments: max. 50 kg

20. Wheels and Tyres: Main landing gear wheel K 525.1 with tyre BARUM 500 x 180 mm
Nose landing gear wheel HP 4808 with tyre BARUM 420 x 150 mm

AIV. Operating and Service Instructions

1. Flight Manual:

- In Czech language: Směrnice pro pilota a mechanika letounu L 200A, Do-L200A-1010.0

2. Maintenance Manual:

- In Czech language: Technická příručka letounů L 200 A, L-200-D, Do-L200AD-1021.0

-In Czech language: Technický popis letounu L 200 A, Do-L200A-1020.0

3. Operational manuals for engines and propellers

-In Czech language: Popis, obsluha a udržování letadlového motoru M 337

-In Czech language: Stavitelná vrtule V 410 A, popis činnosti a ošetření

AV. Notes

1. Both variants of this aircraft were transferred to EASA on Accession of the Czech Republic ('grandfathered').

SECTION A2 Reserved

SECTION B1: GENERAL, L-200 D Type Design

B1. General

Data Sheet No.: EASA.A.043	Issue: 04	Date: 24 May 2006
1. a) Type:	L 200	
b) Variant:	L 200 D	
2. Airworthiness Category:	Normal	
3. Type Certificate Holder::	Aircraft Industries, a.s Na Záhonech 1177 686 04 Kunovice Czech Republic	
4. Manufacturer:	Strojírny první pětiletky n.p. 686 04 Kunovice CZECH REPUBLIC	
5. Certification Application Date:	---	
6. CAA CZ Type Certification Date::	October 12,1962	
7. EASA Type Certification Date	August 12, 2005	
8. The EASA Type Certificate replaces	Czech Type Certificate No.:62001-L200 D	

BII. Certification Basis

1. Reference Date for determining the the applicable requirements	---
2. Certification Basis:	
3. Airworthiness Requirements:	British Civil Airworthiness Requirements (BCAR)normal category, Section D, C (Amendment No. 6 289/62) Civil Air Regulations (CAR) Part 3, normal category, Bauvorschriften für Flugzeuge (BVF) usage group P – transport, loading class 3 – normal
4. Requirements elected to comply:	None
5. EASA Special Conditions:	None
6. EASA Exemptions:	None

7. EASA Equivalent Safety Findings: None
8. EASA Environmental Standards: None

BIII. Technical Characteristics and Operational Limitations

1. Type Design Definition: The specification list of Aircraft L 200 D
Aircraft Drawing No L200.0000
2. Description: Two-engines, five-seater, self-supporting,
low-wing aircraft of all-metal design..
3. Equipment: Flight and navigation instruments:
Airspeed indicator up to 400 km/hr
Altimeter up to 10 km
Variometer
Turn and bank indicator
Combined artificial horizon
Compass
Radio station
Corse gyro
- Engine instruments:
Dual RPM indicator
Dual indicator of filling pressure
Quadruplicate indicator of engine
parameters
Dual indicator of cylinder heads
temperature
- Other instruments:
Landing gear position indicator
Flaps position indicator
Fuel quantity in main tank indicator
Indicator of remaining fuel in tanks
Pressure gauge of hydraulic brakes
Triplicate indicator of heating temperature
Volt-ammeter
Clock
4. Dimensions:
- | | | |
|-----------|--------|----------------|
| Span | 12.325 | m |
| Length | 8.665 | m |
| Height | 2.215 | m |
| Wing Area | 17.280 | m ² |

5. Engines:

5.1. Model:	M 337,
5.2. Type Certificate:	94-06, issued by CAA CZ
5.3. Limitations:	
Maximum take-off power:	
Max. Power	155 kW (210 HP)
Max. Engine speed	2750 RPM
Maximum continuous (nominal) power:	
Max. Power	121 kW (165 HP)
Max. Engine speed	2 600 RPM
Maximum cruising power:	
Max. Power	83,7 kW (132,5 HP)
Max. Engine speed	2400 RPM

6. Propellers:

6.1. Model:	VJ 6.506
6.2. Type Certificate:	6661/61, issued by CAA CZ
6.3. Number of blades:	3
6.4. Diameter:	1 750 mm
6.5. Sense of rotation:	Rotating anti-clock-wise in the view of the flight direction

7. Fluids:

7.1. Fuel	ESSO ICP 80 SHELL Avgas 80 SHELL Avgas 100LL BP 100 L BL 78 according to ČSN 65 6510
7.2. Oil	AEROSHELL Oil W 100 AEROSHELL Oil W 120 ELF Aviation AD 100 MOBIL Aero D 100 BP Aero Oil 100 CASTROL Aero AD 100 TOTAL Aero D100

8. Fluid capacities

8.1. Fuel:	230 litres (2x115 litres) 420 litres (with auxiliary fuel tanks 2x95 l)
8.2. Oil:	2x12 litres

9. Air Speeds:

Manoeuvring Speed	V _A	228	km/h	IAS
Never Exceed Speed	V _{NE}	360	km/h	IAS
Normal operating speed	V _{NO}	280	km/h	IAS
Maximum flaps extended speed	V _{FE}	170	km/h	IAS

10. Maximum Operating Altitude:	5700 m
11. Operational Capability	VFR and IFR flights.
12. Maximum Weights:	
Maximum Weight:	1950 kg
Maximum Weight of non-lifting parts:	1950 kg
13. Centre of Gravity Range:	15,3 % – 29 % M.A.C. (M.A.C.= 1497mm)
14. Datum:	Passes by the fuselage forward leveling items perpendicularly to the aircraft longitudinal axes
15. (reserved)	
16. Levelling Means:	Leveling level is identical with the datum - see leveling diagram shown in the L 200 A Aircraft Maintenance Manual.
17. Minimum Flight Crew:	1 (Pilot)
18. Maximum Passenger Seating Capacity:	4
19. (reserved):	
20. Baggage / Cargo Compartments	max. 50 kg
21. Wheels and Tyres:	Main landing gear wheel K 525.1 with tyre BARUM 500 x 180 mm Nose landing gear wheel HP 4808 with tyre BARUM 420 x 150 mm

BIV. Operating and Service Instructions

1. Flight Manual:

- In Czech language: Směrnice pro pilota letounu L 200D, Do-L200D-1011.0
- In English language: Approved Flight Manual for the MORAVA L 200 D Aeroplane, Do-L200D-1011.2

2. Maintenance Manual:

- In Czech language: Technická příručka letounů L 200 A, L-200-D, Do-L200AD-1021.0
- In English language: Technical Description L 200 D, Do-L200D-1021.2
- In Czech language: Technický popis letounu L 200 A, Do-L200A-1020.0

3. Operational manuals for engines and propellers

- In Czech language: Popis, obsluha a udržování letadlového motoru M 337
- In Czech language: Technický popis a provozní instrukce vrtulové jednotky VJ 6.506
- In Czech language: Technický popis a instrukce pro provoz radiostanice LUN 3521
- In Czech language: Technický popis a instrukce pro obsluhu radiokompasu RKL 301

BV. Notes

None

SECTION B2: Reserved

Change record

Issue 1: Initial issue

Issue 2: Change in address of the TC holder

Issue 3: Change in address of the TC holder

Issue 4: Change in address of the TC holder