



EASA Certification Roadmap on H2 International Workshop

H2 Direct Burn Technology and the specific hazards related to H2 Burn in a gas turbine



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Rolls-Royce Hydrogen Projects

WotAn

LuFo VI, Call 2
Rolls-Royce led



Fundamentals and modelling of H2 combustion

- Simulation
- Experimentation
- Model validation

HyEST

UK ATI
Rolls-Royce led

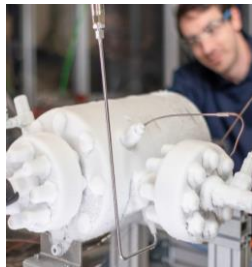


Sub-system architecture and combustion techno.

- Hydrogen combustion
- Methods and tools
- Materials and coating technologies

LH2GT

UK ATI
Rolls-Royce led



System architecture, hydrogen control and fuel system

- LH2 pumping
- Heat management
- H2 valves, actuation & sensing
- Fuel metering and control
- Safety

RACHEL

UK ATI
Rolls-Royce led



Key technologies and integrated powerplant

- Nacelle development
- Supporting systems
- Electrical systems
- Heat exchangers
- Gas tanks
- Aerostructures

CAVENDISH

EU Clean Aviation
Rolls-Royce led



Integrated ground engine testing and requirements for future flight test

- System integration
- Facilities
- Engine test
- Permit to fly study

CONCERTO

EU Clean Aviation
Rolls-Royce supported



Certification req^{TS} to bring new products to market and into service faster

- Future regulations
- Means of compliance

HEAVEN

EU Clean Aviation
Rolls-Royce led



Developing the scalable UltraFan[®] architecture and suite of technologies

- UltraFan technology study
- Hybrid-electric technology



Gas Turbine Technology

Fuel Conditioning

Fuel Management

Combustor

Gas Turbine Operation

Safety

Maintenance Intervals

Refuelling Procedures

Airframe Integration / Aircraft Performance

Fuselage Volume

Fuel Tank Integration

Weight Distribution

Aircraft Operation

Payload/Range Relationship

Tankering & Ferry Range Capabilities

Market & Operational Economics

Test early, trial-fast, learn-by doing with structured de-risk



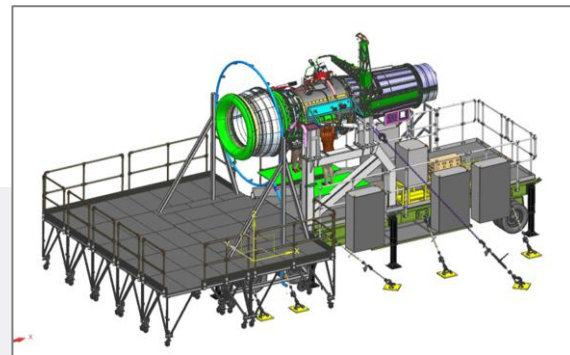
Lavoisier (2022)

- HybAE2100 donor
- Green hydrogen gas demonstration
- Safe fuel handling
- Starting, ignition, shutdown and control stability
- Test at Boscombe Down (UK)



Engine zero (2024)

- Pearl 15 donor with novel hydro-mech- and thermal management system
- Engine oil heat management, coolant: N₂ / hydraulic loop
- VSV actuation
- Combustor fuelled by 100% SAF
- Indoor test at Germany in standard pylon configuration



Engine H2

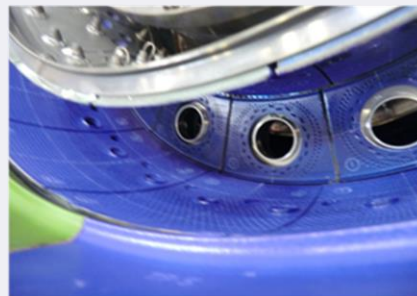
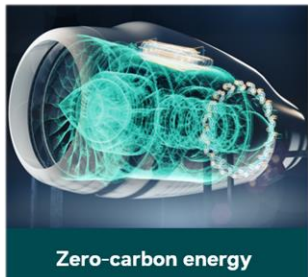
- Fully redressed Pearl 15
- 100% hydrogen distribution-, control- and combustion system
- Integrated engine zero technologies
- Engine handling, operability, maintenance aspects, H₂ safety
- Prove mobile engine test stand and facility
- Outdoors Test at NASA Stennis

H2 Combustion Rig Test Programme

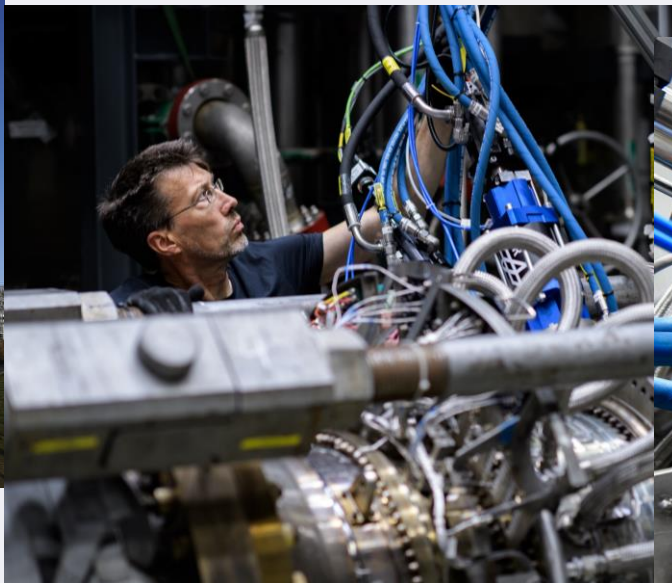
H2 Fuel System Rig

Today's H2 test experience

- Lavoisier green hydrogen gas demonstration
- Engine Zero has proven the hydro/mechanical system and the hydro / gas cooling system. Provided thermal data for more accurate H2 heat management modelling. Demonstrated reliable operability of the novel VSV actuation and control system
- Liquid hydrogen pump test programme commences providing data for engine operability requirements.
- Completion of expansive aero-derivative hydrogen combustion test programme
- Validation on full annular rig at HBK5 2023 & 2024 up to 100% MTO Pearl engine conditions utilizing an unchanged Pearl production combustor with a new H2 fuel injector and H2 fuel manifold (TRL4 achieved)
- *RR Power Systems H2 powered reciprocating engines running already in service*



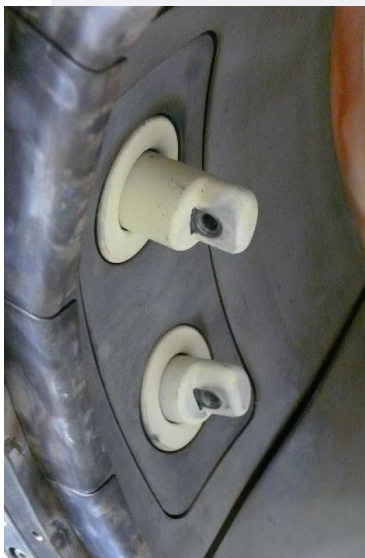
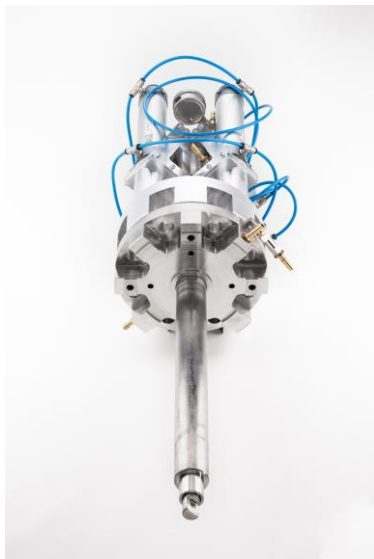
- Leap of insight due to patented DLR technology
- Accelerating decarbonisation of aviation at full pressure conditions [40 bar]



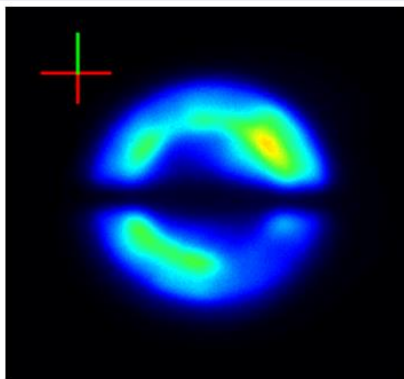
Endoscopic system captures optical data to visualise new combustor technology reaction zones in high TRL facilities

Combination of the unique combustion testing facilities with advanced measurement technologies

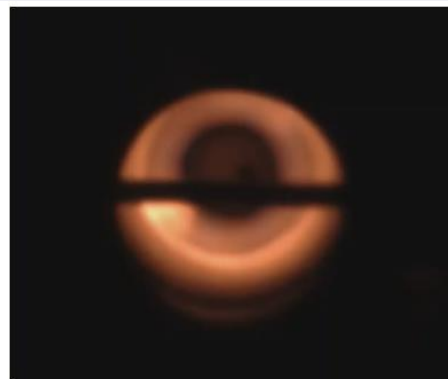
➔ **Measurement technology accelerates the development of low-emission combustion chambers**



For UV (reaction zone) and VIS (hot spots) analysis, the Optical Duct provides additional optical data:



UV visualisation of a combustion oscillation in the H₂/NG mixture



Thermal emission in the VIS of a flashback event at 95 % H₂



Combustion subsystem challenges

- H2 metering and pressure control (Very low H2 mass flows at ignition)
- Transient engine performance (due to 100% efficiency, risk of compressor stall)
- Leakages, embrittlement of fuel system components
- Corrosion. Integrity of ceramic bonding – to be verified
- Whole system integration.

Impact on the engine

- **Ignition:** No issues using existing engine ignition system
- Lean blow-out (Weak extinction): Superior to Jet-A1
- Efficiency: 100% at any condition
- Thermo-acoustics: Similar to Jet-A1 (resonance impact on FMU control to be tested on GH2 engine)
- **Emissions:** All zero, except NOx and increasing quantity of H₂O vapor.
- **Turbine:** Temperature field similar to Jet-A1 (by design), no risks
- **Thermals:** Similar wall temperatures as for Jet-A1, no risks
- **Hardware:** Combustor retrofittable with H2 burners and manifold. Significant changes on the dressed engine (complexity, weight, fire prevention....)
- Altitude **relight:** Superior to Jet-A1 ,based on low TRL test experience (future ALT test scheduled)



EASA help needed

- Definition and requirements of fire zones for GH2 and LH2 applications:

Existing and applicable:

- Designated Fire Zone.
- Fire Zone
- Fuel Leakage Zone

Do we need new definition due to the different types of threats ?

- Fire Withstanding Zone
- Heat Withstanding Zone

Considering different types of threats:

- Micro flames, Torching flames
- Deflagration
- Explosion

- Need of H2 leakage detection / fire detection and fire suppression systems ?
- Standardised means of compliance demonstration
 - Standard H2 burner flame(s) and characteristics
 - Standard H2 fire test(s) procedure
 - Definition of H2 Fireproof and Fire resistant (e.g. Temperature and Heat Flux Density, Duration for different types of flames)



Thank you