



ICAO ENGINE EXHAUST EMISSIONS DATA SHEET

SUBSONIC ENGINES

ENGINE IDENTIFICATION: CFM56-3B-2
UNIQUE ID NUMBER: 1CM005
COMBUSTOR:
ENGINE TYPE: TF
BYPASS RATIO: 5.1
PRESSURE RATIO (π_{00}): 24.1
RATED THRUST (F_{00}) (kN): 98.3

REGULATORY DATA

CHARACTERISTIC VALUE:	HC	CO	NO _x	SMOKE NUMBER
D _p /F ₀₀ (g/kN) or SN	5.3	74.9	49.7	7.7
AS % OF ORIGINAL LIMIT	27.0	63.5	56.4	32.5
AS % OF CAEP/2 LIMIT (NO _x)			70.5	
AS % OF CAEP/4 LIMIT (NO _x)			86.4	
AS % OF CAEP/6 LIMIT (NO _x)			98.2	
AS % OF CAEP/8 LIMIT (NO _x)			118.9	

DATA STATUS

- PRE-REGULATION
- CERTIFICATION
x REVISED (SEE REMARKS)

TEST ENGINE STATUS

x NEWLY MANUFACTURED ENGINES
- DEDICATED ENGINES TO PRODUCTION STANDARD
- OTHER (SEE REMARKS)

EMISSIONS STATUS

x DATA CORRECTED TO REFERENCE
(ANNEX 16 VOLUME II)

CURRENT ENGINE STATUS

(IN PRODUCTION, IN SERVICE UNLESS OTHERWISE NOTED)

x OUT OF PRODUCTION (DATE: -)
- OUT OF SERVICE (DATE: -)

MEASURED DATA

MODE	POWER SETTING (%F ₀₀)	TIME (minutes)	FUEL FLOW (kg/s)	HC	CO	NO _x	SMOKE NUMBER
TAKE-OFF	100	0.7	1.056	0.04	0.90	19.40	6.0
CLIMB OUT	85	2.2	0.878	0.05	0.90	16.70	3.0
APPROACH	30	4.0	0.314	0.07	3.40	8.70	2.5
IDLE	7	26.0	0.119	1.75	30.10	4.10	2.2
LTO TOTAL FUEL (kg) or EMISSIONS (g)			421	337	5988	4213	-
NUMBER OF ENGINES				1	1	1	1
NUMBER OF TESTS				3	3	3	3
AVERAGE D _p /F ₀₀ (g/kN) or AVERAGE SN (MAX)				3.4	61.0	42.9	6.0
SIGMA (D _p /F ₀₀ in g/kN, or SN)				0.4	3.8	0.8	0.8
RANGE (D _p /F ₀₀ in g/kN, or SN)				3.01-3.67	57.6-65.1	42.4-43.9	5.4-6.9

ACCESSORY LOADS

POWER EXTRACTION 0 (kW)
STAGE BLEED 0 (% CORE FLOW)

AT - POWER SETTINGS
AT - POWER SETTINGS

ATMOSPHERIC CONDITIONS

BAROMETER (kPa)	96-97.5
TEMPERATURE (K)	279-286
ABS HUMIDITY (kg/kg)	0.002-0.009

FUEL

SPEC	Jet A
H/C	1.93
AROM (%)	16

MANUFACTURER: CFM International
TEST ORGANIZATION: CFM56 Evaluation Engineering
TEST LOCATION: Peebles Site IVD
TEST DATES: 11/11/1983-14/11/1983

REMARKS

1. Ref GE Report R84AEB245.
2. Engine S/N 692441.
3. Revised based on 3/89 production cycle.
4. Data also apply to CFM56-3C-1 engines rerated to 98.3 kN (22,100 Lb.) thrust for B737-300 and -400 aircraft.

Compliance with Fuel Venting requirements:

- ('x' if complies, 'PR' if pre-regulation, '-' if information is not available)