

TYPE-CERTIFICATE

DATA SHEET

NO. EASA.A.434

for **HB 21**

Type Certificate Holder HB-Flugtechnik

Dr. Adolf Schärfstraße 42 A-4053 Haid Austria

For models: HB 21, HB 21/2400, HB 21/2400 B, HB 21 V1, HB 21 V2



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HB 21

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BAZ/ACG

None

None

SECTION A: HB 21

A.I <u>General</u>

- 1. Type/ Model/ Variant
 - 1.1 Type:

1.2 Model:

- 1.3 Variant:
- Airworthiness Category
 Manufacturer
 B Brditschka Gesmbl Flugzeugbau A-4053 Haid
 - HB Brditschka GesmbH&CoKG Flugzeugbau A-4053 Haid Austria
- 4. EASA Type Certification Application Date
- 5. State of Design Authority
- 6. State of Design Authority Type Certificate Date March 1978 see Note 4
- 7. EASA Type Certification Date

A.II EASA Certification Basis

- 1. Reference Date for determining the applicable requirements
- 2. Airworthiness Requirements LFSM issued 1.November 1975
- 3. Special Conditions
- 4. Exemptions None
- 5. (Reserved) Deviations
- 6. Equivalent Safety Findings
- 7. Environmental Protection
- None Zivilluftfahrzeug-Lärmzulässigkeitverordnung 429/1982

A.III Technical Characteristics and Operational Limitations

1.	Type Desigr	Definition	Drawing Set ar Changes (ÄN	nd following approved [1 – System)	Design	
2.	Description		Single engine, two-seated high wing airplane, wooden wing/steel tube fuselage construction, tandem seating configuration, fixed tri gear, air brakes on upper wing surface and pusher propeller			
3.	Equipment	Minimum Equipment:	1 airspeed indi 1 altimeter wit 1 magnetic cor 1 RPM indicato 1 running time 1 oil pressure g 1 oil temperato 1 Voltmeter 1 fuel pressure 1 fuel quantity 1 stall warning 1 at least 4-poi 1 Masterswitch 1 Currentprote 1 Generator an For Acrobatic i 1 G-Meter 2 Safety loops For Cloudflight 1 Variometer 1 Turn and Bar 1 COM	cator (range up to 250 th mbar barometric dial mpass with deviation ta or meter gauge ure gauge indicator gauge indicator int harness for each sea n ection (circuit protection d 1 Battery n addition for Rudderpedals s hk indicator	km/h) ble ıt ı)	
4.	Dimensions		Span Length Height Wing Area	16,24 m 8,48 m 2,60 m <i>19,00 m</i> ²		
5.	Engine					
	5.1	Model	VW-W-1600	G or G/2		
	5.2	Type Certificate	BAZ TW 2/77			
	5.3	Limitations	Max take-off Max continu	rotational speed ous rotational speed	4000 r.p.m. 3600 r.p.m	
	5.4	Maximum Continuous Power	For power-pl	ants limits refer to Fligh	nt Manual	
6.	Propeller					
	6.1	Model	1 Hoffmann	HO 14 *175 117 LD	or	
				HO 14 *172 117 LD		
	6.2	Type Certificate	LBA 32.110/1	L		



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7	Fluids	6.3	Settings		Low pitch set	tting/ Sta	tic RPM	l:	3400+/- 200
7.	Tulus	7.1	Fuel		AVGAS 100 L Automotive Ga Leaded/unlead (see Note 2)	L or asoline, ded min F	ROZ 98		
		7.1	Oil		quality autor all HD SAE 15V	motive oi V40 (see	ls Castr Flight N	ol GTX2 Ianual)	or
8.	Fluid ca	apaci	ties						
		8.1	Fuel		Standard Fue Total: Usable:	el Tank 54 liter: 53 liter:	s s		
		8.2	Oil		Maximum:	2,5 liter	ſS		
					Minimum:	1,5 liter	ſS		
9.	Air Spe	eds	Design Manoeuvring Sp	eed v _A :		173 km	/h		
		l	Maximum rough air spe	ed Vra)	:	173 km	/h.		
			Never exceed speed v_{NE}	:		200 km	/h		
10.	Approv	ved C	perations Capability		VFR Day				
					Cloud flying pe	ermitted			
11.	Maxim	um N	Vlasses	Taka_of	(see Note 6)			750 ka	
				Maxim	' um mass of nor	n lifting pa	arts	550 kg	
12.	Centre	of G	ravity Range			01		Ū	
			Forward limit				2,410 r	n behind	d Datum
			Rear limit:				2,520 r	n behin	d Datum
13.	Datum				2,00 m in front	t of wing	leading	edge at	root rib 2
14.	Levellir	ng M	eans		top of fuselag	e aft of p	ropelle	r horizoı	ntal
15.	Minim	um F	light Crew		1 (Pilot)				
16.	Maxim	um F	Passenger Seating Capac	city	1				
17.	Baggag	ge/ Ca	argo Compartments		Behind Seats		10 kg		
18.	Wheels	s and	Tyres		Main/Tail Whe For approved	eel Tyre S Types and	ize d rating	see AN	1M

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A.IV Operating and Service Instructions

1.	Flight Manual	Airplane Flight Manual HB 21, Issue 24.March 1983, BAZ approved (German Version)
2.	Maintenance Manual	Maintenance Manual, Issue 24. March 1983, (German Version)
		Engine Manual , Westermayer– VW-V-1600 G
		Hoffmann, Operation and Maintenance Manual for the HOCO propeller, latest Issue or
Ser	vice Informations and Service Bulletins	

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A.V Notes

- 1) Only industrial manufacturing is permitted.
- 2) Use of unleaded automotive fuel SUPER PLUS 98 ÖNorm C1100, min. ROZ 98, in accordance with TM/HB/23/23/93, latest issue, is permitted.
- 3) Modification from engine VW-W 1600-G to G/2 is approved with TM 016
- 4) Initial Certification carried out by the Austrian Aviation Authority Bundesamt für Zivilluftfahrt renamed to Austro Control
- 5) The certification applies to SNo. 21.008 up to 21.029 inclusive.
- 6) The extension of the maximum take off weight from 710 to 750 kg and the maximum weight of the lifting parts from 500 to 550 kg is approved with TM 032/83 dated 13. July 1983, BAZ approved.
- 7) Clowd Flights and Acrobatic is approved in accordance with TM 017/80 dated 1. Sept.1980, BAZ approved.



SECTION B: <u>HB 21/2400</u>

1.1 Type:

B.I General

- 1. Type/ Model/ Variant
- HB 21

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- 1.2 Model:
- 1.3 Variant:
- 2. Airworthiness Category
- 3. Manufacturer

Utility HB Brditschka GesmbH&CoKG Flugzeugbau A-4053 Haid Austria

- 4. EASA Type Certification Application Date
- 5. State of Design Authority BAZ/ACG
- 6. State of Design Authority Type Certificate Date March 1983 see Note 3;
 - The EASA Type Certificate replaces the Austrian Type Certificate SF 2/78

HB 21/2400

7. EASA Type Certification Date

B.II EASA Certification Basis

- 1. Reference Date for determining the applicable requirements
- 2. Airworthiness Requirements LFSM issued 1.November 1975
- 3. Certification Basis LFSM issued 1.November 1975
- 4. Special Conditions
 - 5. Exemptions
- 6. (Reserved) Deviations
- 7. Equivalent Safety Findings
- 8. Environmental Protection
- None None None

Zivilluftfahrzeug-Lärmzulässigkeitverordnung

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B.III Technical Characteristics and Operational Limitations

1.	Type Design	Definition	Drawing Set and following approved Design Changes (ÄM – System)			
2.	Description		Single engine, two-seated high wing airplane, wooden wing/steel tube fuselage construction, tandem seating configuration, fixed tri gear, air brakes on upper wing surface and pusher propeller			
3.	Equipment I	Minimum Equipment:	 1 airspeed indicator (range up to 250 km/h) 1 altimeter with mbar barometric dial 1 magnetic compass with deviation table 1 RPM indicator 1 running time meter 1 oil pressure gauge 1 oil temperature gauge 1 Voltmeter 1 fuel pressure indicator 1 fuel quantity gauge 1 stall warning indicator 1 at least 4-point harness for each seat 1 Masterswitch 1 Current protection (circuit protection) 1 Generator and 1 Battery 			
			1 G-Meter			
			2 Safety loops for Rude	derpedals		
		For Cloudflights	S			
			1 Variometer 1 Turn and Bank indica 2 COM	itor		
4.	Dimensions		Span Length Height Wing Area	16,24 <i>m</i> 8,48 m 2,60 m <i>19,00 m</i> ²		
5.	Engine					
	5.1	Model	VW-HB-2400 G			
	5.2	Type Certificate	BAZ 4/82			
	5.3	Limitations	Max take-off rotation Max continuous rota	nal speed ational speed	4000 r.p.m. 3600 r.p.m	
	5.4	Maximum Continuous Power	For power-plants lim	its refer to Fligh	t Manual	
6.	Propeller					
	6.1	Model	1 Hoffmann HO 14	*175 130 LD o	r	
				HO 14 *172 130	LD	

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		6.2	Type Certificate		LBA 32.110/1	L		
		6.3	Settings		Low pitch set	tting/ Sta	tic RPM	1: 3400+/- 200
7.	Fluids							
		7.1	Fuel		AVGAS 100 L Automotive Ga Leaded/unlead (see Note 4)	L or asoline, ded min F	ROZ 98	
		7.2	Oil		quality autor	notive oi	ls (see l	Flight Manual)
8.	Fluid ca	apaci	ties					
		8.1	Fuel		Standard Fue Total: Usable:	el Tank 54 liters 53 liters	5	
		8.2	Oil		Maximum: Minimum:	4,0 liter 3,0 liter	's 's	
9.	Air Spe	eds						
		I	Design Manoeuvring Sp	beed v _A :		173 km	/h	
		ſ	Maximum rough air spe	eed Vra)	:	173 km	/h.	
		ſ	Never exceed speed v_{NI}	E:		200 km	/h	
10.	Approv	ved O	perations Capability Cloud flying pe	rmitted	VFR Day			
11.	Maxim	um N	lasses		(see Note 6)			
				Take-of	ff	1:6		750 kg
10	Contro	of C	ravity Pango	waximi	um mass of nor	i litting pa	arts	550 Kg
12.	Centre	01 01	Eorward limit				2 /10 n	n behind Datum
			Rear limit:				2,410 m	n behind Datum
13	Datum		Real linnt.		2.00 m in front	t of wing	leading	edge at root rib 2
14.	Levelli	ng Me	ans		top of fuselag	e aft of p	ropeller	r horizontal
15.	Minim	um Fl	ight Crew		1 (Pilot)	e are or p	ropenei	
16.	Maxim	um P	assenger Seating Capa	city	1			
17.	Baggag	ge/ Ca	argo Compartments		Behind Seats		10 kg	
18.	Wheel	s and	Tyres		Main/Tail Whe For approved	eel Tyre S Types and	ize d rating	see AMM



B.IV Operating and Service Instructions

1.	Flight Manual	Airplane Flight Manual HB 21/2400, Issue 15.6. 1983 BAZ approved (German Version)
2.	Maintenance Manual	Maintenance Manual, Issue 24. March 1983, (German Version)
		Engine Manual , VW-HB-2400 G, Oktober 1981 or later approved Issue
Com		Hoffmann, Operation and Maintenance Manual for the HOCO propeller, latest Issue or

Service Informations and Service Bulletins

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B.V <u>Notes</u>

- 1) Only industrial manufacturing is permitted.
- 2) Use of unleaded automotive fuel SUPER PLUS 98 ÖNorm C1100, min. ROZ 98, in accordance with TM/HB/23/23/93, latest issue, is permitted.
- 3) Initial Certification carried out by the Austrian Aviation Authority Bundesamt für Zivilluftfahrt renamed to Austro Control
- 4) The certification applies to SNo. 21.008 up to 21.029 inclusive. The conversion from m HB 21 to variant HB21/2400 is approved.
- 5) The extension of the maximum take off weight from 710 to 750 kg and the maximum weight of the lifting parts from 500 to 550 kg is approved with TM 023/83 dated 13. July 1983, BAZ approved
- 6) Cloud Flights and Acrobatic is approved in accordance with TM 017/80 dated 1. Sept.1980, BAZ approved.
- 7) Glider and Banner towing is approved with TM 021/81 dated 27.2.1982, the following additional equipment must be installed:
 - cylinder head temperature gauge
 Tow indicator in the instrument panel
 coupling type Tost E75
 mirror



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SECTION C: <u>HB 21/2400 B</u>

C.I <u>General</u>

1. Type/ Model/ Variant

1.1 Type:

- HB 21 -
- 1.2 Model:
- 1.3 Variant:
- 2. Airworthiness Category
- 3. Manufacturer

Utility HB Brditschka GesmbH&CoKG Flugzeugbau A-4053 Haid Austria

- 4. EASA Type Certification Application Date
- 5. State of Design Authority BAZ/ACG
- 6. State of Design Authority Type Certificate Date March 1983 see Note 3;
 - The EASA Type Certificate replaces the Austrian Type Certificate SF 2/78

HB 21/2400B

7. EASA Type Certification Date

C.II EASA Certification Basis

- 1. Reference Date for determining the applicable requirements
- Airworthiness Requirements
 Certification Basis
 Special Conditions
 Exemptions
 LFSM issued 1.November 1975
 None
 None
- 6. (Reserved) Deviations
 - 7. Equivalent Safety Findings
- 8. Environmental Protection
- None BAZ approval 6285-2/26-83 dated 25.Juli 1983 Zivilluftfahrzeug-Lärmzulässigkeitverordnung 429/1982



C.III Technical Characteristics and Operational Limitations

1.	Type Design	Definition	Drawing Set and following approved Design Changes (ÄM – System)			
2.	Description		Single engine, two-seated high wing airplane, wooden wing/steel tube fuselage construction, tandem seating configuration, fixed tri gear, air brakes on upper wing surface and pusher propeller			
3.	Equipment I	Minimum Equipment: For Acrobatic ii	 1 airspeed indicator (range up to 250 km/h) 1 altimeter with mbar barometric dial 1 magnetic compass with deviation table 1 RPM indicator 1 running time meter 1 oil pressure gauge 1 oil temperature gauge 1 voltmeter 1 fuel pressure indicator 1 fuel quantity gauge 1 stall warning indicator 1 at least 4-point harness for each seat 1 Masterswitch 1 Current protection (circuit protection) 1 Generator and 1 Battery 			
			1 G-Meter			
		For Claudflight	2 Safety loops for Rud	derpedals		
		For Cloudinght	a 1 Variometer			
			1 Turn and Bank ind	icator		
			1 COM			
4.	Dimensions		Span Length Height Wing Area	16,24 m 8,48 m 2,60 m <i>19,00 m</i> ²		
5.	Engine					
	5.1	Model	VW-HB-2400 G			
	5.2	Type Certificate	BAZ 4/82			
	5.3	Limitations	Max take-off rotation Max continuous rota	nal speed ational speed	4000 r.p.m. 3600 r.p.m	
	5.4	Maximum Continuous Power	For power-plants lim	its refer to Flight	t Manual	
6.	Propeller					
	6.1	Model	1 Hoffmann HO 14	*175 130 LD o	r	
				HO 14 *172 130	LD	



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		6.2	Type Certificate		LBA 32.110/2	1	
		6.3	Settings		Low pitch se	tting/ Static RPI	M: 3400+/- 200
7.	Fluids						
		7.1	Fuel		AVGAS 100 L Automotive Ga Leaded/unlead (see Note 2)	L or asoline, ded min ROZ 98	3
		7.2	Oil		quality autor	motive oils (see	e Flight Manual)
8.	Fluid ca	apaci	ties				
		8.1	Fuel		Standard Fue Total: Usable:	el Tank 54 liters 53 liters	
		8.2	Oil		Maximum: Minimum:	4,0 liters 3,0 liters	
9.	Air Spe	eds					
			Design Manoeuvring S	peed v _A :		173 km/h	
		I	Maximum rough air sp	eed Vra)	:	173 km/h.	
		I	Never exceed speed v_N	IE:		200 km/h	
10.	Approv	ved O	perations Capability		VFR Day		
			Cloud flying pe	rmitted			
11.	Maxim	um N	Aasses				
				Take-of	ff		775 kg
4.2	<u> </u>	()		Maxim	um mass of nor	n lifting parts	550 kg
12.	Centre	of G	ravity Range			2 440	un habinal Datum
			Forward limit			2,410	m benind Datum
12	Datum		Rear limit:		2 00 m in from	2,520 t of wing loadin	m benind Datum
13. 14	Lovollin		0.205		2,00 mmmmul	t of wing leading	or horizontal
14. 15	Minim	um E	eans light Crow		1 (Pilot)	e art of propert	
15. 16	Maxim	um P	assenger Seating Cana	acity	1		
17	Raggag	onn n re/ Ca	argo Compartments	icity	- Rehind Seats	10 kg	
18.	Wheel	s and	Tyres		Main/Tail Whe	eel Tyre Size Types and ratin	g see AMM



C.IV Operating and Service Instructions

1.	Flight Manual	Airplane Flight Manual HB 21/2400B, Issue 24.3. 1983 BAZ approved (German Version)
2.	Maintenance Manual	Maintenance Manual, Issue 24. March 1983, (German Version)
		Engine Manual , VW-HB-2400 G, Oktober 1981 or later approved Issue
		Hoffmann, Operation and Maintenance Manual for the HOCO propeller, latest Issue or Service Informations and Service Bulletins

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C.V <u>Notes</u>

- 1) Only industrial manufacturing is permitted.
- 2) Use of unleaded automotive fuel SUPER PLUS 98 ÖNorm C1100, min. ROZ 98, in accordance with TM/HB/23/23/93, latest issue, is permitted.
- 3) Initial Certification carried out by the Austrian Aviation Authority Bundesamt für Zivilluftfahrt renamed to Austro Control
- 4) The certification applies to SNo. 21.008 up to 21.029 inclusive.
- 5) Cloud Flights and Acrobatic is approved in accordance with TM 017/80 dated 1. Sept.1980, BAZ approved.
- 6) Glider and Banner towing is approved with TM 021/81 dated 27.2.1982, the following additional equipment must be installed:
 - 1 cylinder head temperature gauge 1 Tow indicator in the instrument panel 1 coupling type Tost E75 1 mirror



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HB 21 VI

SECTION D: <u>HB 21 V1</u>

D.I General

- 1. Type/ Model/ Variant
 - 1.1 Type: HB 21
 - 1.2 Model:
 - 1.3 Variant:
- 2. Airworthiness Category
- 3. Manufacturer

Utility HB Brditschka GesmbH&CoKG Flugzeugbau A-4053 Haid Austria

- 4. EASA Type Certification Application Date
- 5. State of Design Authority BAZ/ACG
- 6. State of Design Authority Type Certificate Date 16 August 1978 see Note 2;
 - The EASA Type Certificate replaces the Austrian Type Certificate SF 4/78
- 7. EASA Type Certification Date

D.II EASA Certification Basis

- 1. Reference Date for determining the applicable requirements
- 2. Airworthiness Requirements LFSM issued 1.November 1975
- 3. Certification Basis LFSM issued 1.November 1975
- 4. Special Conditions
- 5. Exemptions
- 6. (Reserved) Deviations
- 7. Equivalent Safety Findings
- 8. Environmental Protection
- None None None
- Zivilluftfahrzeug-Lärmzulässigkeitverordnung 486/1972



D.III Technical Characteristics and Operational Limitations

1.	L. Type Design Definition			Drawing Set and following approved Design Changes (ÄM – System)				
2.	Descrip	tion		Single engine, two-seated high wing airplane, wooden wing/steel tube fuselage construction, tandem seating configuration, fixed tri gear, air brakes on upper wing surface and pusher propeller				
3.	8. Equipment Minimum Equipment:			 airspeed indicator (range up to 250 km/h) altimeter with mbar barometric dial magnetic compass with deviation table RPM indicator running time meter oil pressure gauge oil temperature gauge Voltmeter fuel pressure indicator fuel quantity gauge stall warning indicator at least 4-point harness for each seat Masterswitch Current protection (circuit protection) Generator and 1 Battery 				
4.	Dimens	ions		Span Length Height Wing Area	16,24 <i>m</i> 8,48 m 2,60 m 19,00 m²			
5.	Engine							
		5.1	Model	VW-HB-1600 G	ì			
		5.2	Type Certificate	BAZ 2/77				
		5.3	Limitations	Max take-off ro Max continuo	otational speed us rotational speed	4000 r.p.m. 3600 r.p.m		
		5.4	Maximum Continuous Power	For power-plar	nts limits refer to Fligh	t Manual		
6.	Propelle	er						
		6.1	Model	1 Hoffmann H	HO 14 *175 117			
		6.2	Type Certificate	LBA 32.110/1				
		6.3	Settings	Low pitch setti	ng/Static RPM:	3400+/- 200		
7.	Fluids	0.0		Low pitch setti		0.000, 200		
-		7.1	Fuel	AVGAS 100 LL o Automotive Gase Leaded/unleade (see Note 1)	or oline, d min ROZ 98			
		7.2	Oil	quality automo	otive oils Castrol GTX2	or		



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all HD SAE 15W40 (see Flight Manual)

8.	Fluid capa	cities					
	8.1	Fuel		Standard Fue	l Tank		
				lotal: Usable:	54 liters	5	
	8.2	Oil		Maximum [.]	2.5 liter	, ,	
	0.2			Minimum:	1,5 liter	S	
9.	Air Speeds	i de la companya de l					
		Design Manoeuvring Sp	peed v _A :		160 km,	/h	
		Maximum rough air spe	eed Vra):		160 km,	/h.	
		Never exceed speed v_{NI}	E:		175 km,	/h	
10	A						
10.	Approved	Operations Capability		VFR Day			
11.	Maximum	Masses	(Take-off	see Note 6)			661 kg
			Maximu	m mass of non	lifting pa	arts	467 kg
12.	Centre of (Gravity Range			01		C C
		Forward limit				2,410 m	n behind Datum
		Rear limit:				2,520 m	n behind Datum
13.	Datum		Ĩ	2,00 m in front	of wing	leading	edge at root rib 2
14.	Levelling N	leans	t	op of fuselage	e aft of p	ropeller	horizontal
15.	Minimum	Flight Crew	-	L (Pilot)			
16.	Maximum	Passenger Seating Capa	city 2	L			
17.	Baggage/ (Cargo Compartments	E	Behind Seats		10 kg	
18.	Wheels an	d Tyres	1 F	Main/Tail Whe For approved T	el Tyre S Types and	ize 1 rating	see AMM



D.IV Operating and Service Instructions

1.	Flight Manual	Airplane Flight Manual HB 21/2400, Issue 16.8. 1978 BAZ approved (German Version)
2.	Maintenance Manual	Maintenance Manual, Issue 16.8. 1978 (German Version)
		Engine Manual , VW-W 1600 G Hoffmann, Operation and Maintenance Manual for the HOCO propeller, latest Issue or Service

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Informations and Service Bulletins



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D.V <u>Notes</u>

- 1) Use of unleaded automotive fuel SUPER PLUS 98 ÖNorm C1100, min. ROZ 98, is permitted.
- 2) Initial Certification carried out by the Austrian Aviation Authority Bundesamt für Zivilluftfahrt renamed to Austro Control
- 3) The certification applies to SNo. 21.001



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HB 21 V2

SECTION E: <u>HB 21 V2</u>

E.I <u>General</u>

- 1. Type/ Model/ Variant
 - 1.1 Type: HB 21
 - 1.2 Model:
 - 1.3 Variant:
- 2. Airworthiness Category
- 3. Manufacturer

Utility HB Brditschka GesmbH&CoKG Flugzeugbau A-4053 Haid Austria

- 4. EASA Type Certification Application Date
- 5. State of Design Authority BAZ/ACG
- 6. State of Design Authority Type Certificate Date 10 April 1978 see Note 2;
 - The EASA Type Certificate replaces the Austrian Type Certificate SF 3/78
- 7. EASA Type Certification Date

E.II EASA Certification Basis

- 1. Reference Date for determining the applicable requirements
- 2. Airworthiness Requirements LFSM issued 1.November 1975
- 3. Certification Basis LFSM issued 1.November 1975
- 4. Special Conditions
 - 5. Exemptions
- 6. (Reserved) Deviations
- 7. Equivalent Safety Findings
- 8. Environmental Protection
- None None None
- Zivilluftfahrzeug-Lärmzulässigkeitverordnung 486/1972



E.III Technical Characteristics and Operational Limitations

1.	Type Desig	n Definition	Drawing Set and following approved Design Changes (ÄM – System)			
2.	Description Single engine, two-seated high wing airplane, wooden wing/steel tube fuselage construction tandem seating configuration, fixed tri gear, ai brakes on upper wing surface and pusher prop				rplane, truction, gear, air ner propeller	
3.	Equipment	: Minimum Equipment:	 1 airspeed indicator (range up to 250 km/h) 1 altimeter with mbar barometric dial 1 magnetic compass with deviation table 1 RPM indicator 1 running time meter 1 oil pressure gauge 1 oil temperature gauge 1 Voltmeter 1 fuel pressure indicator 1 fuel quantity gauge 1 stall warning indicator 1 at least 4-point harness for each seat 1 Masterswitch 1 Current protection (circuit protection) 1 Generator and 1 Battery 			
4.	Dimension	S	Span Length Height Wing Area	16,24 m 8,48 m 2,60 m 19,00 m ²		
5.	Engine		0	,		
	5.1	Model	VW-W 1600 G			
	5.2	Type Certificate	BAZ 2/77			
	5.3	Limitations	Max take-off rotatior Max continuous rota	al speed ational speed	4000 r.p.m. 3600 r.p.m	
	5.4	Maximum Continuous Power	For power-plants lim	its refer to Fligh [,]	t Manual	
6.	Propeller					
	6.1	Model	1 Hoffmann HO 14	*175 117		
	6.2	Type Certificate	LBA 32.110/1			
	6.3	Settings	Low pitch setting/ Sta	atic RPM:	3400+/- 200	
7.	Fluids 7.1	Fuel	AVGAS 100 LL or Automotive Gasoline, Leaded/unleaded min (see Note 1)	ROZ 98		
	7.2	Oil	quality automotive o all HD SAE 15V	ils Castrol GTX2 N4 (see Flight N	or Ianual)	



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8.	Fluid capacities						
	8.1	Fuel		Standard Fue Total: Usable:	l Tank 54 liters 53 liters	5	
	8.2	Oil		Maximum: Minimum:	2,5 liter 1,5 liter	'S 'S	
9.	Launching	Hooks					
10.	Air Speeds						
	·	Design Manoeuvring Sp	peed v _A :		166 km	/h	
		Maximum rough air spo	eed Vra):		166 km	/h.	
		Never exceed speed $v_{\mbox{\tiny N}}$	e:		200 km	/h	
11.	Approved (Operations Capability		VFR Day			
12.	Maximum	Masses					
			Take-off				661 kg
			Maximu	m mass of non	lifting pa	arts	467 kg
13.	Centre of C	Gravity Range					
		Forward limit				2,410 m	n behind Datum
		Rear limit:				2,520 m	n behind Datum
14.	Datum		ź	2,00 m in front	of wing	leading	edge at root rib 2
15.	Levelling N	leans	t	top of fuselage	e aft of p	ropeller	horizontal
16.	Minimum F	light Crew	-	1 (Pilot)			
17.	Maximum	Passenger Seating Capa	city 1	1			
18.	Baggage/C	Cargo Compartments	E	Behind Seats		10 kg	
19.	Wheels and	d Tyres	r F	Main/Tail Whe For approved 1	el Tyre S Types and	ize d rating	see AMM



E.IV Operating and Service Instructions

1.	Flight Manual	Airplane Flight Manual HB 21/2400, Issue 10.April 1978, BAZ approved (German Version)
2.	Maintenance Manual	Maintenance Manual, Issue 10.April 1978 (German Version)
		Engine Manual , VW-W 1600 G,
		Hoffmann, Operation and Maintenance Manual for the HOCO propeller, latest Issue or
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Service Informations and Service Bulletins

All Master Manuals are issued in German Language only



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Notes

- 1) Use of unleaded automotive fuel SUPER PLUS 98 ÖNorm C1100, min. ROZ 98, is permitted.
- 2) Initial Certification carried out by the Austrian Aviation Authority Bundesamt für Zivilluftfahrt renamed to Austro Control
- 3) The certification applies to SNo. 21.002 up to 21.007 including



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SECTION F: ADMINISTRATIVE SECTION

F.I Acronyms & Abbreviations

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F.II <u>Type Certificate Holder Record</u> HB-Flugtechnik GmbH Dr. Adolf Schärfstraße 42 A-4053 Haid, Austria <u>www.hb-flugtechnik.at</u>

F.III Change Record

Issue	Date	Changes	TC Issue No. & Date
Issue 01	25 January	Initial Issue; Transfer from ACG TCDS SF 2/78 issue 5, SF 4/78	Initial Issue,
	2010	issue 1 and SF 3/78 issue 1 to the EASA Type Design	25 January
			2010
Issue 02	08 January	New EASA template; Correction of Maximum Passenger	Initial Issue,
	2024	Seating Capacity for each variant;	25 January
			2010

-END-

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