



EASA

European Aviation Safety Agency

Continuing Airworthiness of Type Design – EU Products Large Transport Aeroplanes

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Introduction





Agenda

- ICAO Continued Airworthiness Definition
- State of Design Responsibilities
- Regulations
- Occurrence Reporting
- Definition of Unsafe Condition
- Airworthiness Directive
 - Compliance Time Determination
- Differences with other FAA system



➤ ICAO Doc N° 9760-2001 defines the Continued Airworthiness as:

- 'The processes that ensure, at anytime in it's life, an aircraft complies with the technical conditions fixed to the issue of the Certificate of Airworthiness and is in a condition for safe operation.'

➤ and recommends:

- 'Contracting states are required to have a system that... ensures aircraft are in a condition for safe operation.'

**EU and National Regulations on CAW
are built on this OACI recommendation**



State of Design Responsibilities

➤ As per ICAO Annex 8 Chapter 4

- The State of Design of an aircraft shall transmit to every Contracting State any generally applicable information which it has found necessary for the CAW of an aircraft, including its engines and propellers when applicable, and for the safe operation of the aircraft (mandatory continuing airworthiness information – MCAI) and notification of suspension or revocation of a Type Certificate.
- The term “MCAI” is intended to include mandatory requirements for modification, replacement of parts or inspection of aircraft and amendment of operating limitations and procedures. Among such information is that issued by Contracting States in the form of airworthiness directives.

MCAI = AD



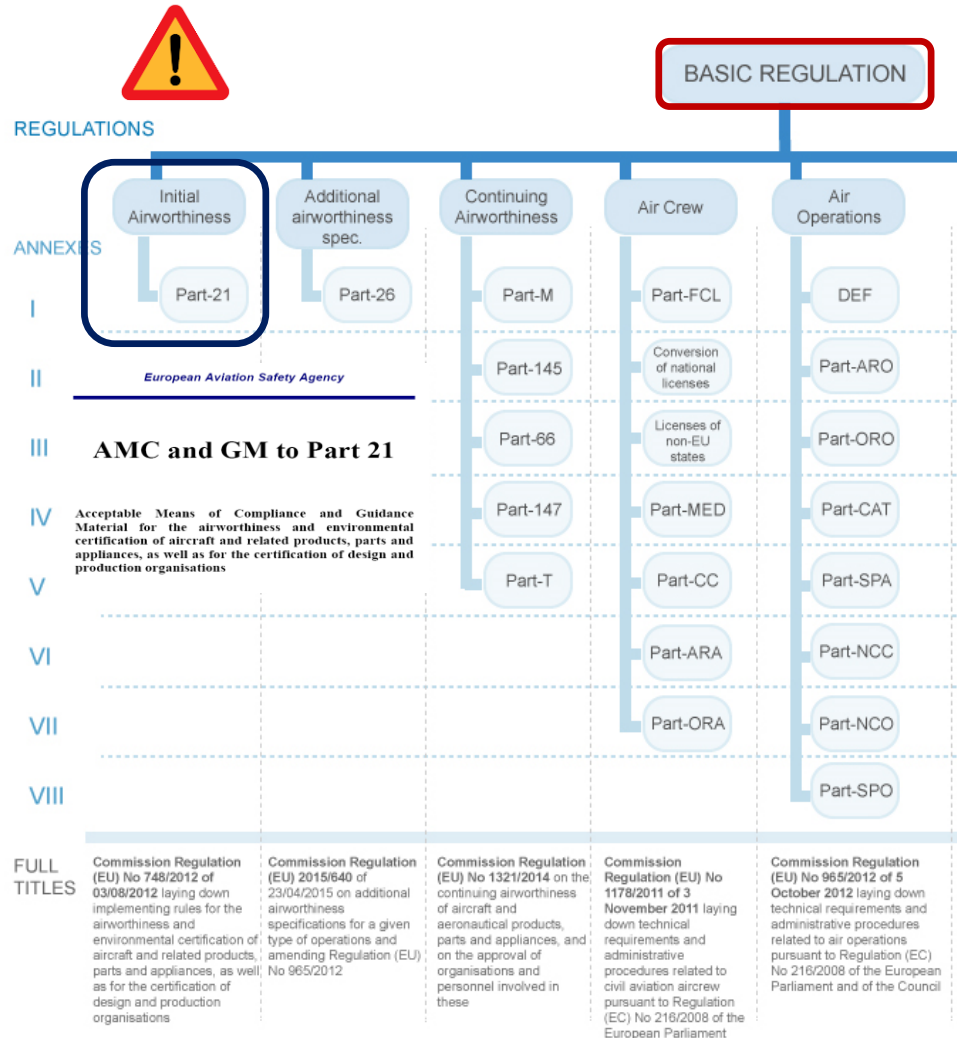
Continued Airworthiness Scope

- Continuing Airworthiness covers all the processes ensuring that all aircraft comply with the airworthiness requirements in force and are in condition for safe operation, at any time during their operating life
- CAW of Type Design is one of those processes



EU CAW Legal Framework

Regulations Structure



➤ Article 20 – Airworthiness and Environmental Certification

- Transfer of SoD tasks
- React without undue delay to a safety problem and issue/disseminate the applicable mandatory information

➤ Part 21

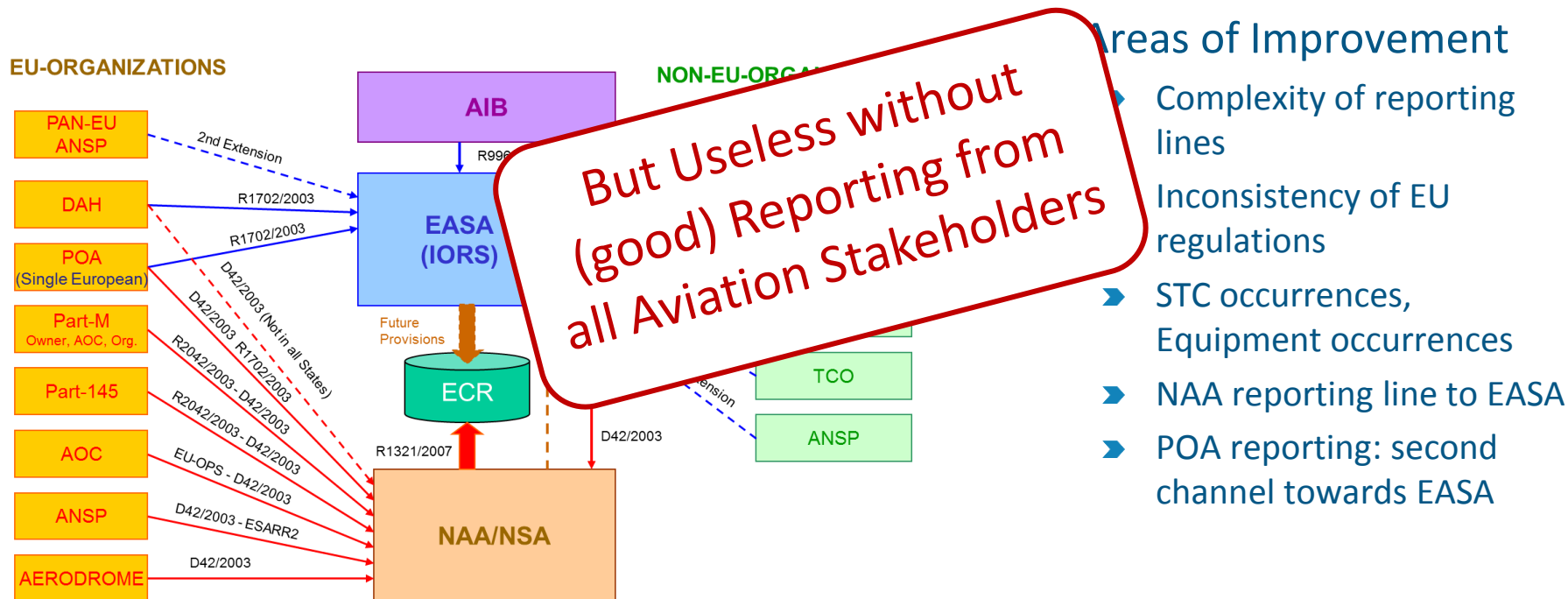
- 21A.3 Failures, malfunctions and defects
 - Obligations of DAH
- 21A.3B Airworthiness Directives
 - Obligations of DAH and EASA
- Associated AMC and GM, and AMC 20-28



Occurrence Reporting Lines

► Occurrence Reporting Lines

► ICAO Annex 8, EU Directive 2003/42/EC, BR Article 15, Part 21, Part M, OPS regulations,...





➤ Has resulted in or may result in an unsafe condition

➤ Results of new test, updated analysis






Part 21 – Annex 1

21A.3 - Failures, malfunctions and defects

TC/STC Holders Must

21A.3 Failures, malfunctions and defects

- (a) System for Collection, Investigation and Analysis of Data
- Information about this system shall be made available to all known operators
- (b) Reporting to the Agency.
1. The holder shall report to the Agency any failure, malfunction, defect or other occurrence 72 hours after the identification of the possible unsafe condition 
 2. These report hours after
- (c) Investigation of Reported Occurrences.
1. When an occurrence reported under paragraph (b). or under 21A.129(f)(2) or 21A.165(f)(2) results from a deficiency in the design, or a manufacturing approval, ETSO approval, or any other relevant approval deemed to have been issued under this Regulation, and which has resulted in or may result in an unsafe condition, shall investigate the reason
 2. If the Agency finds that an action is required to correct the deficiency, the holder of the type-certificate, restricted type-certificate, supplemental type-certificate, major repair design approval, ETSO authorisation, or any other relevant approval deemed to have been issued under this Regulation, or the manufacturer as appropriate, shall submit the relevant data to the Agency.



Definition of Unsafe Condition

AMC 21.A.3B(b) Unsafe condition

An unsafe condition exists if there is **factual evidence** (Factual service experience, analysis or tests) that:



- (a) An event may occur that would result in **fatalities**, usually with the **loss of the aircraft**, or **reduce the capability of the aircraft** or the **ability of the crew** to cope with adverse operating conditions to the extent that there would be:
- (i) A large reduction in safety margins or functional capabilities, or
 - (ii) Physical distress or excessive workload such that the flight crew cannot be relied upon to perform their tasks accurately or completely, or
 - (iii) Serious or fatal injury to one or more occupants

unless it is shown that the event is within the limit defined by the applicable Certification Basis elements, or



Definition of Unsafe Condition

AMC 21.A.3B(b) Unsafe condition

- (b) There is an unacceptable risk of serious or **fatal injury to persons other than occupants**, or 
- (c) **Design features** intended to minimise the effects of **survivable accidents** are **not performing their intended function**. 



AMC 21.A.3B(b) - Notes

AMC 21.A.3B(b) Unsafe condition (cont'd)

Note 1: **Non-compliance** with applicable regulatory requirements is generally considered as an unsafe condition. **Non-Compliance** in that possible events resulting from this non-compliance **do not constitute an unsafe condition** as defined under paragraphs (a), (b) and (c).

Note 2: An unsafe condition is different from non-compliance. **Unsafe different than compliance** **airworthiness requirements are considered**

Note 3: The above definition covers the **majority of cases** where the Agency considers there is a **Definition not exhaustive, overriding consideration may exist**, **considerations may lead the Agency to issue an airworthiness directive.**

Note 4: There may be cases where **events can be considered as an unsafe condition if they occur too frequently** (significantly beyond the applicable safety objectives) and could eventually lead to consequences listed in paragraph (a) in **specific operating environment**. **Need to reconcile Safety objectives and real failure case rate** listed in paragraph (a), the referenced events may reduce the capability of the aircraft or the ability of the crew to cope with adverse operating conditions to the extent that there would be, for example, a **significant reduction in safety margins or functional capabilities, a significant increase in crew workload, or in conditions impairing crew efficiency, or discomfort to occupants, possibly including injuries.**



CS25.1309/AMC25.1309 - Classification

Effect on airplane	No effect on operational capabilities or safety	Slight reduction in functional capabilities or safety margins	Significant reduction in functional capabilities or safety margins	Large reduction in functional capabilities or safety margins	Normally with hull loss
Effect on occupants excluding flight crew	Inconvenience	Physical discomfort	Physical distress, possibly including injuries	Serious or fatal injury to a small number of passengers or cabin crew	Multiple fatalities
Effect on flight crew	No effect on flight crew	Slight increase in workload	Physical discomfort or a significant increase in workload	Physical distress or excessive workload which impairs ability to perform tasks	Fatalities or incapacitation
Allowable qualitative probability	No probability requirement	Probable	Remote	Extremely remote	Extremely improbable
Allowable quantitative average probability per FH	No probability requirement	$<10^{-3}$	$<10^{-5}$	$<10^{-7}$	$<10^{-9}$
Classification of failure conditions	No safety effect	Minor	Major	Hazardous	Catastrophic



Definition of Unsafe Conditions – Usual Issues

- Who is responsible for making the determination?
 - TC/STC Holders
- CAT or HAZ events, importance of probability
- MAJOR events, classification and probability
 - Frequency
 - Interpretation of « *overriding safety considerations* »
- Which criteria for “risk to non-occupants”
 - 1kg, large parts, runway safety
- Unsafe condition with no aircraft malfunction
- Crashworthiness issues



Corrective action(s)

➤ GM 21A.3B(d)(4)

- To be used to determine AD compliance time
- Recognises the need to manage risk
- Recognises the need to maintain aviation services
- First need to restore adequate level of risk through inspection, limitations
- Method not intended to avoid shorter reaction times
- Upper limit for probability level, $2 \cdot 10^{-6}$ /FH for CAT
- Fleet criteria is also included (0,1 for CAT)



Corrective action(s)

➤ GM 21A.3B(d)(4) Defect Correction

➤ Different steps to be considered for hazardous to catastrophic failure conditions:

- i - Establish **all possible alleviating action** such as **Risk Alleviation** crew drills, route restrictions, and other limitations
- ii - **Identify that part of the fleet,** **Impacted Fleet** to the residual risk
- iii - **Using reasonably cautious assumptions,** calculate the likely hazardous/catastrophic **rate** **Individual Risk** **carrying the risk** in the affected fleet
- iv - Compare **Proposed Campaign Risk Acceptability** campaign will correct the deficiency with the time suggested (fig.2 of GM)
- v - Also ensure that the expected probability of the catastrophic event during the rectification period on **Fleet Risk** **is in accordance with Fleet risk** (fig.4 of GM)



GM 21A.3B(d)(4) – Assumptions Individual Risk

Occurrence Classification

If Probability above – Fleet grounded

Catastrophic

Maximum Allowance per aircraft:

1 in 10^7 flight hours

Hazardous

Maximum Allowance per aircraft:

1 in 10^5 flight hours

75% of this allowance is considered to be “basic design risk”.
That leaves only 25% for unforeseen campaign situations

Maximum Allowance per aircraft:

$$25\% * 1 * 10^{-7} = 0.25 * 10^{-7}$$

Maximum Allowance per aircraft:

$$25\% * 1 * 10^{-5} = 0.25 * 10^{-5}$$

10 Rectification Campaigns

Maximum Allowance per aircraft per campaign:

$$\frac{0.25 * 10^{-5}}{10} = 0.25 * 10^{-6}$$

Maximum Allowance per aircraft per campaign:

$$\frac{0.25 * 10^{-7}}{10} = 0.25 * 10^{-8}$$

Aircraft Design Life – 60,000 FH

Event Level per campaign:

$$60\,000 * 0.25 * 10^{-8} = 1.5 * 10^{-4}$$

Event Level per campaign:

$$60\,000 * 0.25 * 10^{-6} = 1.5 * 10^{-2}$$

Probability of Occurrence for a given Failure Condition (PO): $PO * RT1 \leq \text{Event Level per campaign}$

Reaction Time

Reaction Time 1:

$$RT1 = \frac{1.5 * 10^{-4}}{PO}$$

Reaction Time 1:

$$RT1 = \frac{1.5 * 10^{-2}}{PO}$$



GM 21A.3B(d)(4) – Assumptions Fleet Risk

Occurrence Classification

Catastrophic

Large fleet criterion: 0.1

Hazardous

Large fleet criterion: 0.5

For a given Failure Condition (PO) and a number of affected A/C (N): $PO * RT2 * N \leq \text{Large fleet criterium}$

Maximum average reaction time for the fleet

Reaction Time RT2

$$RT2 = \frac{0.1}{N * PO}$$

Reaction Time RT2

$$RT2 = \frac{0.5}{N * PO}$$

Threshold of fleet size criteria becoming
predominant

N=666

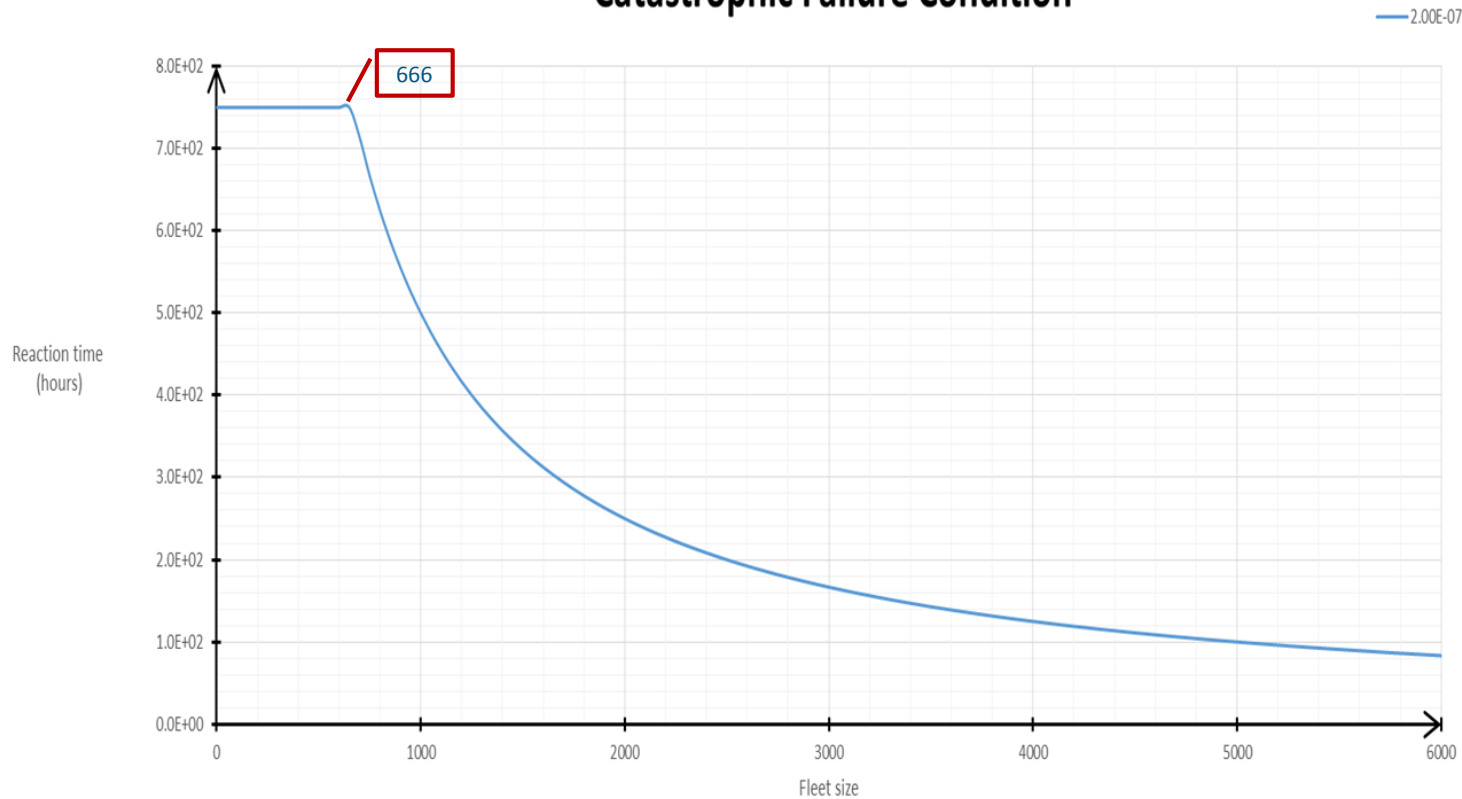
N=33

**WATHEVER THE METHOD, IT GIVES THE MAXIMUM AVERAGE
REACTION TIME for THE WHOLE AFFECTED FLEET**



Example – $P_0=2.10^{-7}$ ADL=60,000FH

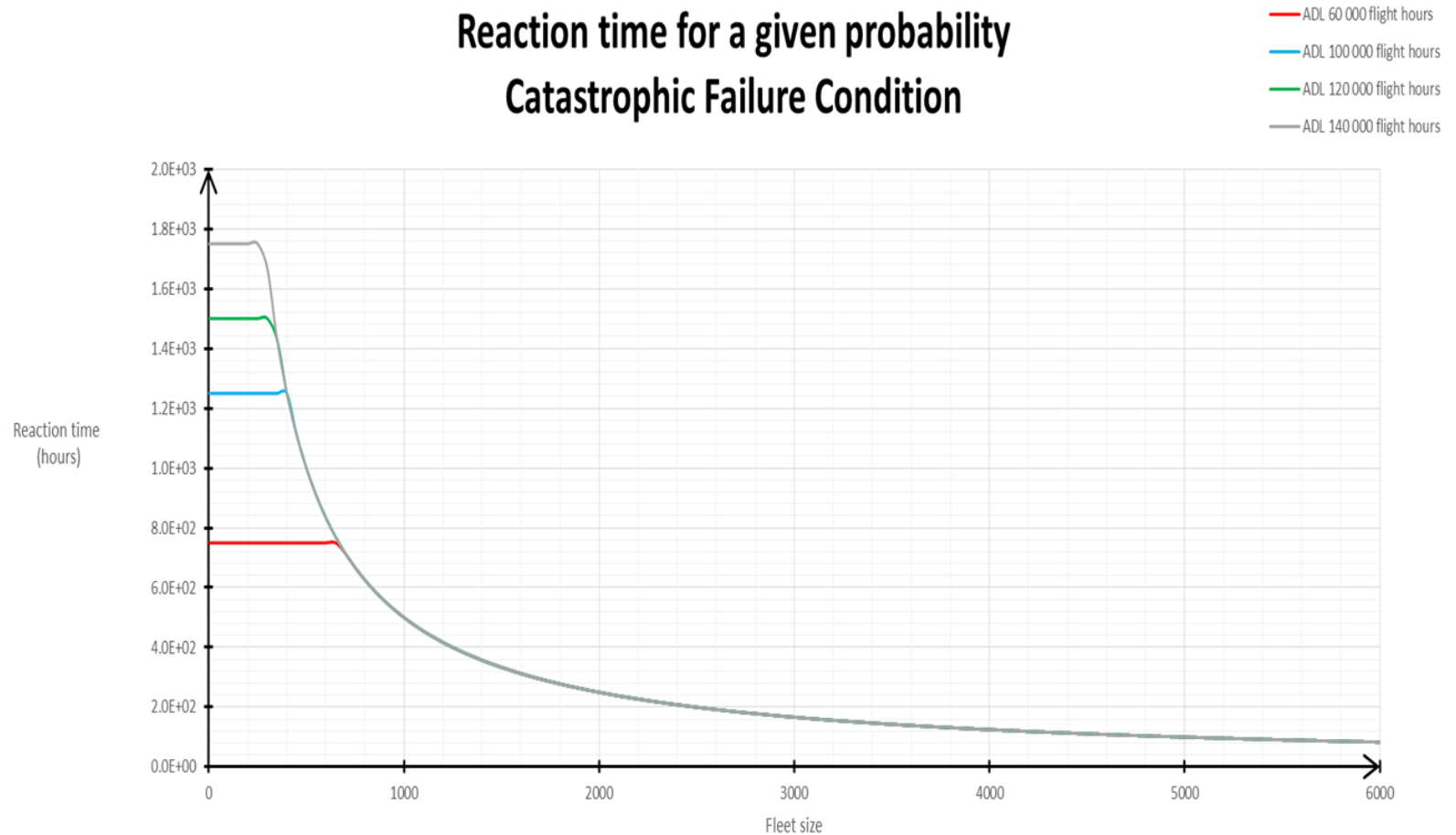
Reaction time for a given probability Catastrophic Failure Condition





Effect of the Design Service Life

Reaction time for a given probability Catastrophic Failure Condition





Reaction Time Table

Estimated catastrophe rate to aircraft due to the defect under consideration (per a/c hour)	Average reaction time for aircraft at risk (hours)	On a calendar basis
4×10^{-8}	3 750	15 months
5×10^{-8}	3 000	12 months
1×10^{-7}	1 500	6 months
2×10^{-7}	750	3 months
5×10^{-7}	300	6 weeks
1×10^{-6}	150	3 weeks
1×10^{-5}	15	Return to base



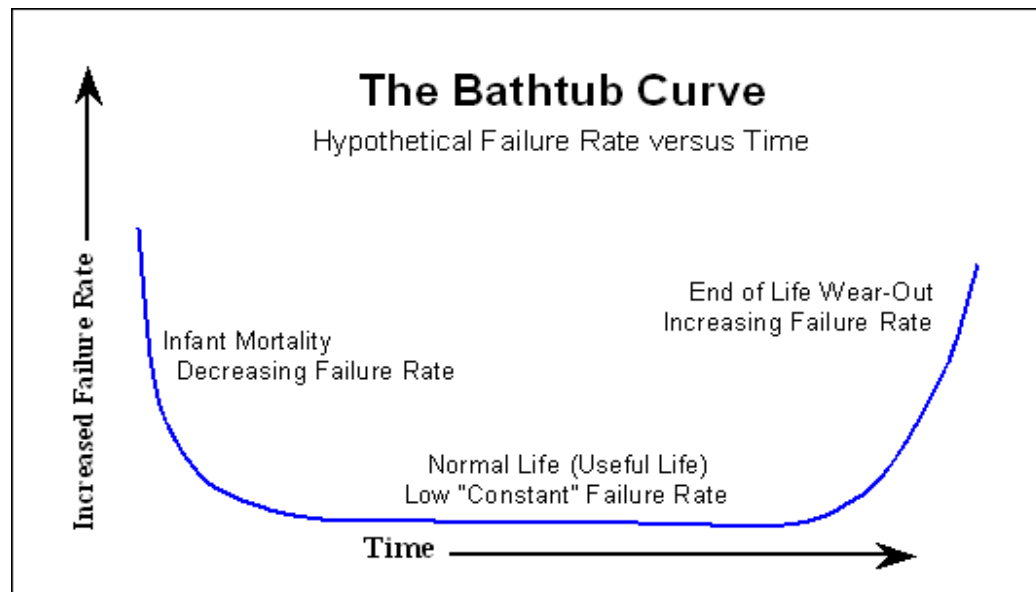
However...

- The Guidance Material
 - Is a Guidance!
 - Is not intended to avoid quicker reaction times without high expense or disruption of service
- A considerable amount of engineering judgement is necessary to take into account the multiple real life factors
- Final Decision may be tempered by non numerical considerations



Other Consideration

- The Guidance may not be adequate to tackle
 - Aging issues – Old A/C prioritisation
 - Infant mortality issues – new A/C prioritisation
 - Fatigue – depend on FC and FH considerations





Typical Issues encountered

- TCH/STCH tend to use the method for all issues
 - As described, not always adequate
- Not necessarily intended to be used for structure issues and for safety features
- Need to be cautious in estimation of probability
 - Some fancy statistical approaches
- When to start the clock?
 - Date of the Unsafe Condition, of the occurrence...
 - SB availability

USE COMMON SENSE



Decision making

- Team: PCM with experts – primary level
 - Identify Unsafe Condition and discuss corrective actions
 - Together with TC/STC Holder
- Product Line Section Manager
 - Endorsement (or not) of the proposal
 - Check and agree PAD
 - Sign AD
- Escalation Process
 - Large Aeroplane Safety Board
 - CT management
 - ISC



Difference with FAA System - TARAM

- TARAM =Transport Airplane Risk Assessment Methodology
- TARAM Methodology
 - Meant to calculate risks and not really compliance times
 - Part of a method also requiring engineering judgment
 - Is more complex than the Part 21 GM
 - Cost related



TARAM – The tool – Risks Description

► 5 different risks

- **Uncorrected Individual Risk:** “probability of individual fatal injury per flight hour, during a reasonable number of flights that occur any time in the remaining life of affected airplanes, if no mandatory action(s) is required”
- **Uncorrected Fleet Risk:** “number of severity-weighted events due to a specific condition in the remaining life of affected airplanes if no mandatory action is required”
- **90-Day Fleet Risk:** “number of fatal injuries expected to occur due to a specific condition in affected airplanes in the 90 days following the determination that a specific condition is a safety issue”
- **Control Program Individual Risk:** “probability of individual fatal injury per flight hour occurring during a reasonable number of flights within the control program time”
- **Control Program Fleet Risk:** “number of fatal injuries (or severity-weighted events) expected due to a specific condition in affected airplanes during the control program time”



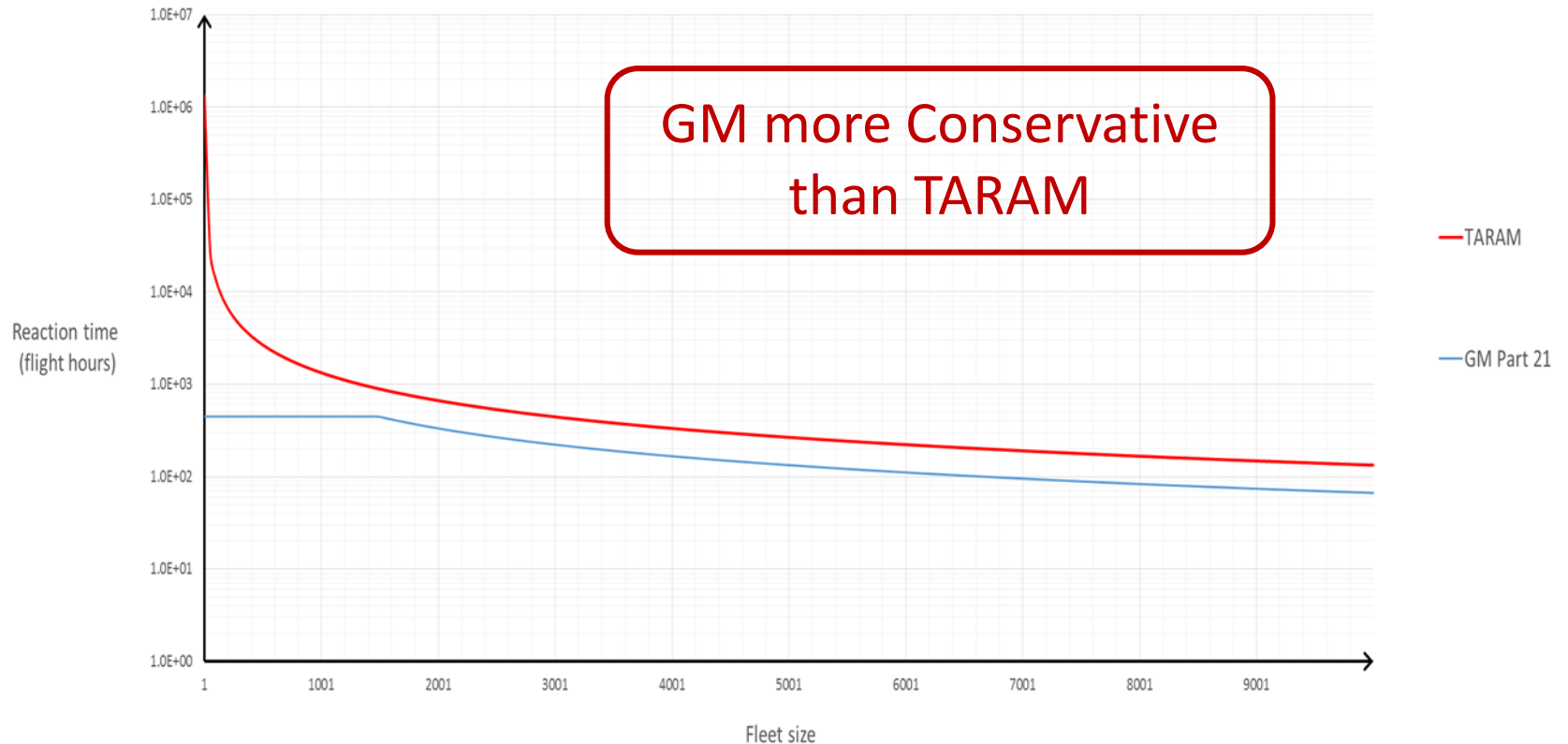
Example - Hypothesis

- A passenger A/C (TARAM guideline = 0.02)
- $PO = 1,5 \cdot 10^{-7}$ per flight hour
- Utilization $U = 2.1$ hours per day i.e. an Aircraft Design Life of 27 000 flight hours flying 35 years
- 10 campaigns
- A catastrophic failure condition
- $CP \cdot IR = 0.1$ (Comparison 1)
- $CP \cdot IR = 0.2$ (Comparison 2)
- $CP \cdot IR = 0.34$ (Comparison 3)



Comparison 1

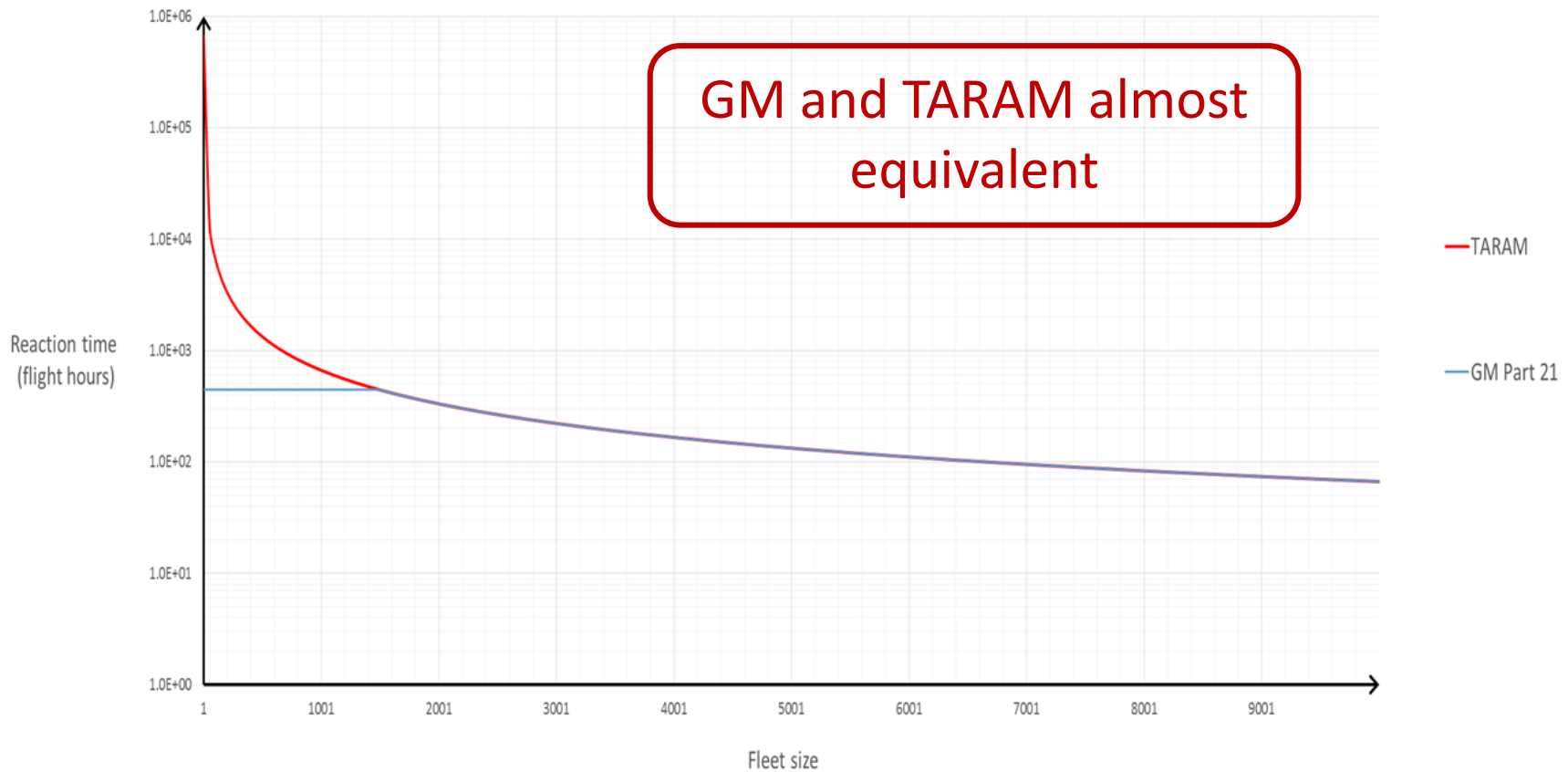
Comparison between TARAM and GM 21.A.3B(d)(4) compliance times
 $CP*IR < (CP*IR)_{min}$





Comparison 2

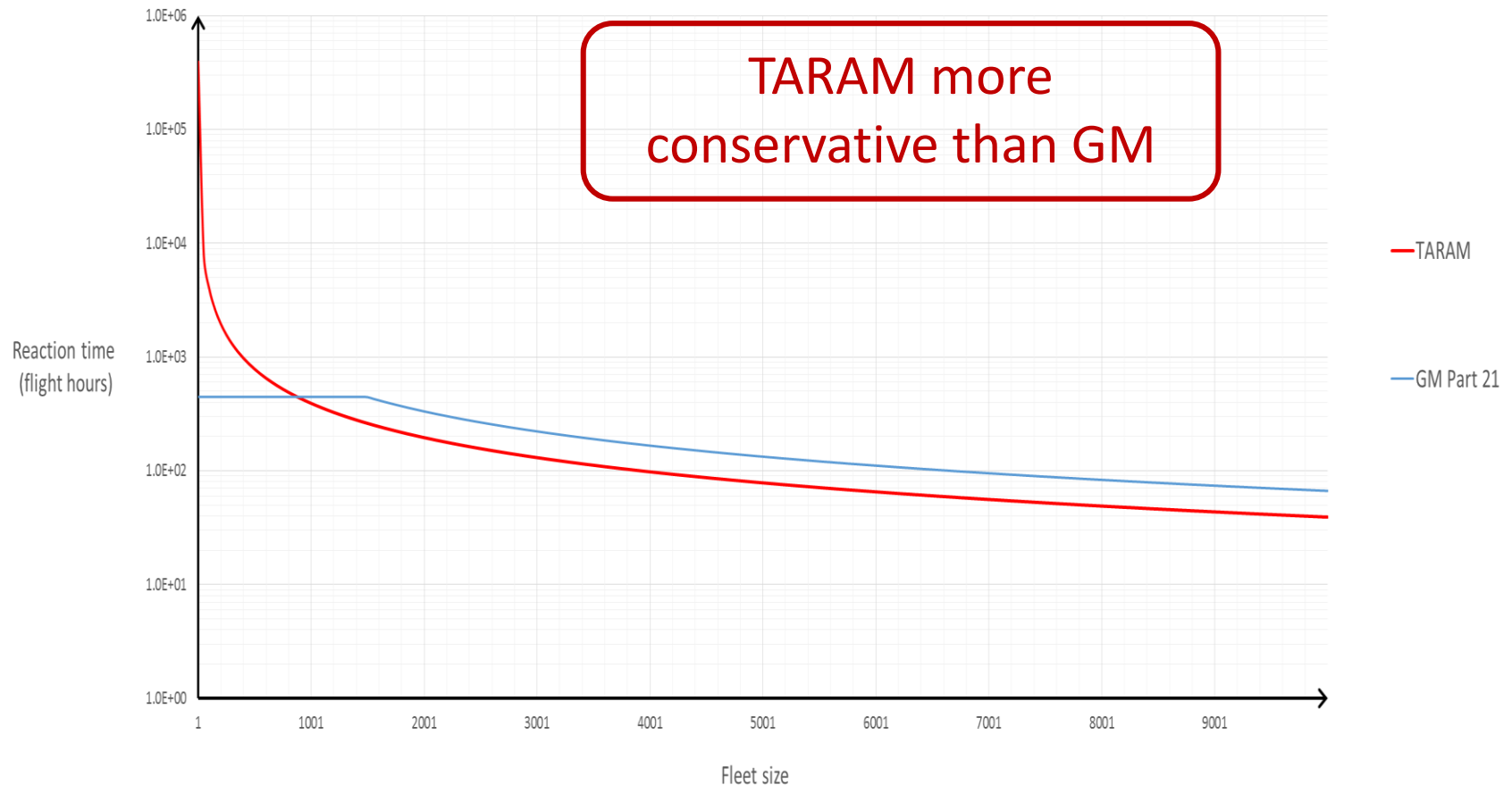
Comparison between TARAM and GM 21.A.3B(d)(4) compliance times
 $CP*IR = (CP*IR)_{min}$





Comparison 3

Comparison between TARAM and GM 21.A.3B(d)(4) compliance times
 $CP*IR > (CP*IR)_{min}$





Conclusions

- The Guidance is a starting point
- It does not represent the reality
 - A whole set of assumptions are behind them and each TC/STC holder must be aware of them
 - Is not a magic tool
 - Require extensive and specialised knowledge to be used
- Different assessment methodologies exist
 - Rather consistent
 - Complexity vs simplicity
- Engineering Judgement and Common Sense are a MUST for correct CAW assessment



Conclusions

- Determination of Unsafe Condition and Compliance Time is
 - Following a structured process
 - Following a structured decision making
 - Always focusing on Safety



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**BUT Experience shows it provides
adequate Level of Safety
Up to now...**

**NO Compromise with
Safety**

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**Thank You
Any Question?**

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