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# TYPE-CERTIFICATE DATA SHEET

NO. EASA.A.358

**for**  
BÖLKOW BO 208

**Type Certificate Holder**  
Airbus Defence and Space GmbH

Willy-Messerschmitt-Straße 1  
82024 Taufkirchen  
Germany

For models: Bölkow Junior  
Bölkow BO 208 C Junior



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<b>SECTION A: BÖLKOW JUNIOR</b> .....	<b>4</b>
<b>A.I. General</b> .....	<b>4</b>
<b>A.II. EASA Certification Basis</b> .....	<b>4</b>
<b>A.III. Technical Characteristics and Operational Limitations</b> .....	<b>5</b>
<b>A.IV. Operating and Service Instructions</b> .....	<b>7</b>
<b>A.V. Notes</b> .....	<b>7</b>
<b>SECTION B: BÖLKOW BO 208 C JUNIOR</b> .....	<b>8</b>
<b>B.I. General</b> .....	<b>8</b>
<b>B.II. EASA Certification Basis</b> .....	<b>8</b>
<b>B.III. Technical Characteristics and Operational Limitations</b> .....	<b>9</b>
<b>B.IV. Operating and Service Instructions</b> .....	<b>11</b>
<b>B.V. Notes</b> .....	<b>11</b>
<b>SECTION ADMINISTRATIVE</b> .....	<b>12</b>
<b>I. Acronyms &amp; Abbreviations</b> .....	<b>12</b>
<b>II. Type Certificate Holder Record</b> .....	<b>12</b>
<b>III. Change Record</b> .....	<b>12</b>



## **SECTION A: BÖLKOW JUNIOR**

### **A.I. General**

1. Type/ Model/ Variant	
1.1 Type	Bölkow BO 208
1.2 Model	Bölkow Junior
1.3 Variant	N/A
2. Airworthiness Category	Normal Utility
3. Manufacturer	Bölkow-Apparatebau GmbH Werk Laupheim Nabern/Teck, Württ., Germany  Waggon- und Maschinenbau AG Siebelwerke ATG GmbH Donauwörth, Germany
4. EASA Type Certification Application Date	15 October 2014 (see note 2)
5. State of Design Authority	Germany (see note 2)
6. State of Design Authority Type Certificate Date	22 April 1963 (see note 2)
7. EASA Type Certification Date	02 February 2015 (see note 2)

### **A.II. EASA Certification Basis**

1. Reference Date for determining the applicable requirements	N/A
2. Airworthiness Requirements	CAR Part 3 dated 15 May 1956 plus Amendment 3-1 through 3-7
3. Special Conditions	N/A
4. Exemptions	N/A
5. (Reserved) Deviations	N/A
6. Equivalent Safety Findings	N/A
7. Environmental Protection	ICAO Annex 16, Vol. I; for details see TCDSN.A.358



### **A.III. Technical Characteristics and Operational Limitations**

1. Type Design Definition	Set of drawings, specifications and reports
2. Description	Single engine, strut-braced shoulder-wing two place aircraft with nose wheel configuration, metal construction
3. Equipment	Required equipment acc. to certification standard CAR Part 3 Additionally: 1 stall warning and 2 shoulder harnesses
4. Dimensions	Wing Span: 7,42m or 8,02m (TM 208-22/64) Length: 5,79m Height: 1,98m
5. Engine	
5.1. Model	Engine 1: Continental O-200-A Engine 2: Continental RR O-200-A
5.2 Type Certificate	Engine 1: EASA.IM.E.101 Engine 2: US E3IN
5.3 Limitations	Maximum continuous speed 2750rpm
6. Load factors	Normal: $n = -1,52$ to $+3,8$ Utility: $n = -1,76$ to $+4,4$
7. Propeller	
7.1 Model	Propeller 1: Hoffmann Propeller GmbH & Co. KG HOCO F-H2/LC14-170 123 7R Propeller 2: McCauley 1A100MCM 6758 Propeller 3: McCauley 1A100MCM 6950 Propeller 4: McCauley 1A100MCM 6955 Note: Propellers 1 through 4 can alternatively be fitted to engines 1 and 2
7.2 Type Certificate	Propeller 1: DE 32.110/1 Propeller 2-4: US P-918
7.3 Number of blades	2
7.4 Diameter	Propeller 1 & 2: 170cm Propeller 3 & 4: 175cm
7.5 Sense of Rotation	Clockwise
8. Fluids	
8.1 Fuel	Aviation fuel, min. 80/87 octane
8.2 Oil	below $+4^{\circ}\text{C}$ ( $40^{\circ}\text{F}$ ) SAE 30 above $+4^{\circ}\text{C}$ ( $40^{\circ}\text{F}$ ) SAE 50
8.3 Coolant	N/A
9. Fluid capacities	
9.1 Fuel	Max fuel quantity: 80l or 100l (TM 208-18/64) Usable fuel quantity: 77,5l or 97,5l
9.2 Oil	4,7l
9.3 Coolant system capacity	N/A



10. Air Speeds	Never Exceed Speed	$V_{NE}$	153kts
	Manoeuvring Speed	$V_A$	106kts
	Maximum Normal Operating Speed	$V_{NO}$	124kts
	Maximum Flap Extended Speed	$V_{FE}$	79kts
11. Flight Envelope	Not specified		
12. Approved Operations Capability	VFR Day, no icing		
13. Maximum Masses	Maximum Take-off mass		
	Normal:	600kg	
	Utility:	575kg	
14. Centre of Gravity Range	Normal:	Max. FWD:	1711mm @ 532kg linear to 1770mm @ 600kg
		Max. AFT:	1792mm @ 600kg linear to 1807mm @ 520kg
	Utility:	Max. FWD:	1711mm @ 532kg linear to 1746mm @ 575kg
		Max. AFT:	1777mm @ 575kg linear to 1807mm @ 520kg
15. Datum	Reference plane is 1900mm FWD of reference point (red encircled rivet on each fuselage side wall, located 100mm FWD of hole center of lower wing strut fitting)		
16. Control surface deflections	Aileron	Up	25° (+/-1°)
		Down	12° (+/-1°)
	Rudder	Left/Right	20° (+/-2°)
	Elevator	Up	18° (+/-1°)
		Down	9° (+/-1°)
	Trim tab	see Maintenance Manual	
	Flaps	Up	0°
		Down	35° (+/-1°)
17. Levelling Means	horizontal using extended line from level marks on left side of fuselage		
18. Minimum Flight Crew	1		
19. Maximum Passenger Seating Capacity	1		
20. Baggage/ Cargo Compartments	max. 20kg		
21. Wheels and Tyres	5.00 x 5"		
22. (Reserved)			



#### **A.IV. Operating and Service Instructions**

- |                                |   |
|--------------------------------|---|
| 1. Flight Manual               | Flight Manual Bölkow Junior, DVL/PfL-approved 15 February 1963 (KF-1D or KF-1E) incl. revisions   |
| 2. Maintenance Manual          | Operator's Handbook Bölkow Junior, date of issue November 1962 (KF-2D or KF-2E) incl. revisions<br>Maintenance Manual Bölkow Junior, date of issue November 1962 (KF-3D or KF-3E) incl. revisions |
| 3. Structural Repair Manual    | Not specified   |
| 4. Weight and Balance Manual   | Covered in Flight Manual, Operator's Manual and Maintenance Manual  |
| 5. Illustrated Parts Catalogue | Ersatzteilliste Bölkow Junior, date of issue October 1964 incl. revisions   |

#### **A.V. Notes**

1. Serial Numbers 500 to 566
2. The EASA TCDS is based on the LBA TCDS No. 644/SA for BO 208 (at Issue 12, dated 12 April 2005)



## **SECTION B: BÖLKOW BO 208 C JUNIOR**

### **B.I. General**

1. Type/ Model/ Variant	
1.1 Type	Bölkow BO 208
1.2 Model	Bölkow BO 208 C Junior
1.3 Variant	N/A
2. Airworthiness Category	Normal Utility
3. Manufacturer	Bölkow-Apparatebau GmbH Werk Laupheim Nabern/Teck, Württ., Germany  Waggon- und Maschinenbau AG Siebelwerke ATG GmbH Donauwörth, Germany
4. EASA Type Certification Application Date	15 October 2014 (see note 2)
5. State of Design Authority	Germany (see note 2)
6. State of Design Authority Type Certificate Date	20 May 1965 (see note 2)
7. EASA Type Certification Date	02 February 2015 (see note 2)

### **B.II. EASA Certification Basis**

1. Reference Date for determining the applicable requirements	N/A
2. Airworthiness Requirements	CAR Part 3 dated 15 May 1956 plus Amendment 3-1 through 3-7
3. Special Conditions	N/A
4. Exemptions	N/A
5. (Reserved) Deviations	N/A
6. Equivalent Safety Findings	N/A
7. Environmental Protection	ICAO Annex 16, Vol. I; for details see TCDSN.A.358





### **B.III. Technical Characteristics and Operational Limitations**

1. Type Design Definition	Set of drawings, specifications and reports
2. Description	Single engine, strut-braced shoulder-wing two place aircraft with nose wheel configuration, metal construction
3. Equipment	Required equipment acc. to certification standard CAR Part 3 Additionally: 1 stall warning and 2 shoulder harnesses
4. Dimensions	Wing Span: 8,02m Length: 5,79m Height: 1,98m
5. Engine	
5.1. Model	Engine 1: Continental O-200-A Engine 2: Continental RR O-200-A
5.2 Type Certificate	Engine 1: EASA.IM.E.101 Engine 2: US E3IN
5.3 Limitations	Maximum continuous speed 2750rpm
6. Load factors	Normal: $n = -1,52$ to $+3,8$ Utility: $n = -1,76$ to $+4,4$
7. Propeller	
7.1 Model	Propeller 1: McCauley 1A100MCM 6758 Propeller 2: McCauley 1A100MCM 6950 Propeller 3: McCauley 1A100MCM 6955 Note: Propellers 1 through 3 can alternatively be fitted to engines 1 and 2
7.2 Type Certificate	Propeller 1-3: US P-918
7.3 Number of blades	2
7.4 Diameter	Propeller 1: 170cm Propeller 2 & 3: 175cm
7.5 Sense of Rotation	Clockwise
8. Fluids	
8.1 Fuel	Aviation fuel, min. 80/87 octane
8.2 Oil	below $+4^{\circ}\text{C}$ ( $40^{\circ}\text{F}$ ) SAE 30 above $+4^{\circ}\text{C}$ ( $40^{\circ}\text{F}$ ) SAE 50
8.3 Coolant	N/A
9. Fluid capacities	
9.1 Fuel	Max fuel quantity: 100l Usable fuel quantity: 97,5l
9.2 Oil	4,7l
9.3 Coolant system capacity	N/A



10. Air Speeds	Never Exceed Speed	$V_{NE}$	153kts
	Manoeuvring Speed	$V_A$	106kts
	Maximum Normal Operating Speed	$V_{NO}$	124kts
	Maximum Flap Extended Speed	$V_{FE}$	79kts
11. Flight Envelope	Not specified		
12. Approved Operations Capability	VFR Day, no icing		
13. Maximum Masses	Maximum Take-off mass		
	Normal:	630kg	
	Utility:	600kg	
14. Centre of Gravity Range	Normal:	Max. FWD:	1710mm @ 574kg linear to 1760mm @ 630kg
		Max. AFT:	1800mm @ 630kg linear to 1825mm @ 550kg
	Utility:	Max. FWD:	1710mm @ 574kg linear to 1732mm @ 600kg
		Max. AFT:	1810mm @ 600kg linear to 1820mm @ 550kg
15. Datum	Reference plane is 1900mm FWD of reference point (red encircled rivet on each fuselage side wall, located 100mm FWD of hole center of lower wing strut fitting)		
16. Control surface deflections	Aileron	Up	25° (+/-1°)
		Down	12° (+/-1°)
	Rudder	Left/Right	20° (+/-2°)
	Elevator	Up	18° (+/-1°)
		Down	9° (+/-1°)
	Trim tab	see Maintenance Manual	
	Flaps	Up	0°
		Down	35° (+/-1°)
17. Levelling Means	horizontal using extended line from level marks on left side of fuselage		
18. Minimum Flight Crew	1		
19. Maximum Passenger Seating Capacity	1		
20. Baggage/ Cargo Compartments	max. 20kg		
21. Wheels and Tyres	Main Tyres 5.50 x 5" or 380 x 150 (TM208-1/84)		
	Nose Tyre 5.00 x 5"		
22. (Reserved)			



#### **B.IV. Operating and Service Instructions**

- |                                |   |
|--------------------------------|---|
| 1. Flight Manual               | Flight Manual Bölkow Bo 208 C Junior, DVL/PfL-approved 19 May 1965 (KIM-60D or KIM-60E) incl. revisions   |
| 2. Maintenance Manual          | Operator's Handbook Bölkow Bo 208 C Junior, date of issue April 1965 (KIM-61D) or August 1965 (KIM-61E) incl. revisions<br>Maintenance Manual Bölkow Bo 208 C Junior, date of issue April 1965 (KIM-62D) or August 1965 (KIM-62E) incl. revisions |
| 3. Structural Repair Manual    | Not specified   |
| 4. Weight and Balance Manual   | Covered in Flight Manual, Operator's Manual and Maintenance Manual  |
| 5. Illustrated Parts Catalogue | Spare Parts Catalogue Bölkow Bo 208 C Junior, date of issue September 1966 (BI99D) or June 1967 (BI99E) incl. revisions   |

#### **B.V. Notes**

1. Serial Numbers 567 and higher
2. The EASA TCDS is based on the LBA TCDS No. 644/SA for BO 208 C Junior at Issue 11, dated 12 April 2005



## **SECTION ADMINISTRATIVE**

### **I. Acronyms & Abbreviations**

CAR	Civil Aviation Regulations
DVL/PfL	Deutsche Versuchsanstalt für Luftfahrt / Prüfstelle für Luftfahrtgerät
FAA	Federal Aviation Administration
FWD	Forward
ICAO	International Civil Aviation Organization
LBA	Luftfahrt Bundesamt
N/A	Not applicable
SAE	Society of Automotive Engineers
TCDS	Type Certificate Data Sheet
VFR	Visual Flight Rules

### **II. Type Certificate Holder Record**

<b>Day of Entry</b>	<b>Company Name (Legal Entity)</b>
26.06.1958	Bölkow Apparatebau GmbH
11.07.1969	Messerschmitt-Bölkow-Blohm GmbH
01.04.1992	Messerschmitt-Bölkow-Blohm AG
30.09.1992	Deutsche Aerospace AG
02.01.1995	Daimler-Benz Aerospace AG
17.11.1998	Daimler Chrysler Aerospace AG
10.07.2000	EADS Deutschland GmbH
01.07.2014	Airbus Defence and Space GmbH

### **III. Change Record**

<b>Issue</b>	<b>Date</b>	<b>Changes</b>	<b>TC Issue No. &amp; Date</b>
01	02 February 2015	Initial Issue after TC transfer	01, 02 February 2015
02	06 March 2015	References to initial LBA TCDS in A.I.7 and B.I.7 corrected	01, 02 February 2015
03	22 June 2015	Type Certificate Holder Record revised	01, 02 February 2015
04	13 November 2018	Change of TC holder address	02, 13 November 2018
05	15 October 2020	Minor updates to wording and content	02, 13 November 2018

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