



TYPE-CERTIFICATE DATA SHEET

NO. EASA.A.576

for
P2010 TYPE

Type Certificate Holder
Costruzioni Aeronautiche TECNAM SPA
Via S. D'acquisto,62
80042 Boscotrecase (Na)
ITALIA

For models: P2010, P2010 TDI



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SECTION A: P2010

A.I. General

1. Type/Model/Variant
 - 1.1 Type P2010
 - 1.2 Model: P2010
 - 1.3 Variant --
2. Airworthiness Category: CS-23 Normal category
3. Manufacturer: see A.VI, note 5
4. EASA Type Certification Application Date: 15 September 2010
5. State of Design Authority N/A
6. State of Design Authority Type Certificate Date N/A
7. EASA Type Certification Date 26 September 2014

A.II. EASA Certification Basis

1. Reference Date for determining the applicable requirements: 15 September 2010
2. Airworthiness Requirements: EASA CS-23 amdt.2 dated 28 September 2010
EASA CS-ACNS
3. Special Conditions: CRI B-52 (SC-B23.div-01 Human Factors – Integrated Avionic System);
CRI F-101 (SC-F23-1309-02 Protection from the Effect of HIRF);
CRI F-54 (SC-F23-1309-03 Protection from the Effects of Lightning Strike, Indirect Effects);
CRI F-58 (SC-F23.1353-02 Lithium Battery Installations)
- see *Explanatory Note TCDS EASA.A.576*
4. Exemptions: None
5. Deviations: None
6. Equivalent Safety Findings: None
7. Requirements elected to comply: EASA CS-23 amdt.4 para. 23.1306
EASA CS-23 amdt.4 para. 23.1308
8. Environmental Protection: Refer to TCDSN EASA.A.576



A.III. Technical Characteristics and Operational Limitations

1. Type Design Definition: Document no. 2010/010 "Type Design Definition"
2. Description:
 - 2.1 Basic: Single-engine, fixed pitch propeller, four seats, high wing aeroplane equipped with fixed tricycle landing gear, featuring composite, aluminium and steel construction.
 - 2.2 Optional (see A.VI, note 1,3) Single-engine, variable pitch propeller, four seats, high wing aeroplane equipped with fixed tricycle landing gear, featuring composite, aluminium and steel construction.
3. Equipment: Equipment list, AFM, doc. No. 2010/100, Section 6
4. Dimensions:

Span	10.30 m (33.79 ft)
Length	7.97 m (26.15 ft)
Height	2.64 m (8.66 ft)
	2.84 m (9.32 ft) (see A.VI, note 8)
Wing Area	13.9 m ² (149.6 ft ²)
5. Engine:
 - 5.1 Basic
 - 5.1.1 Model: Lycoming Engines: IO-360-M1A
 - 5.1.2 Type Certificate: EASA Type Certificate No. EASA.IM.E.032
 - 5.1.3 Limitations
 - 5.1.3.1 Basic: Take-Off Power 134 kW (180HP) at 2700 RPM
Max continuous power 134 kW (180HP) at 2700 RPM
Other engine's limitations are listed in doc. No. 2010/100 "P2010 Aircraft Flight Manual", Section 2
 - 5.1.3.2 Optional (see A.VI, note 1) Take-Off Power 134 kW (180HP) at 2700 RPM
Max continuous power 129 kW (173HP) at 2600 RPM
Other engine's limitations are listed in doc. No. 2010/100 "P2010 Aircraft Flight Manual", Section 2
 - 5.2 Optional (see A.VI, note 3)
 - 5.2.1 Model: Lycoming Engines: IO-390-C3B6
 - 5.2.2 Type Certificate: EASA Type Certificate No. EASA.IM.E.097
 - 5.2.3 Limitations: Take-Off Power 160.3 kW (215HP) at 2700 RPM
Max continuous power 160 kW (215HP) at 2700 RPM
Other engine's limitations are listed in doc. No. 2010/100 "P2010 Aircraft Flight Manual", Section 2
6. Load factors:

	Flap UP	Flap DOWN
Positive	+3.8 g	+2.0 g
Negative	-1.52 g	0.0 g
7. Propeller:
 - 7.1 Basic:
 - 7.1.1 Model: MT Propeller: MT 188 R 145-4G



- 7.1.2 Type Certificate: EASA Type Certificate No. EASA.P.006
- 7.1.3 Number of blades: 2
- 7.1.4 Diameter: 1.880 m (74 in) – No reduction is permitted
- 7.1.5 Sense of Rotation: Clockwise (pilot’s view)
- 7.2 Optional 1:
(see A.VI, note 1)
- 7.2.1 Model: MT Propeller: MTV-15-B/193-52 () (see A.VI, note 6)
- 7.2.2 Type Certificate: EASA Type Certificate No. EASA.P.098
- 7.2.3 Number of blades: 2
- 7.2.4 Diameter: 1.930 m (76 in) – No reduction is permitted
- 7.2.5 Sense of Rotation: Clockwise (pilot’s view)
- 7.3 Optional 2:
(see A.VI, note 3)
- 7.3.1 Model: MT Propeller: MTV-12B/183-59 () (see A.VI, note 6)
- 7.3.2 Type Certificate: EASA Type Certificate No. EASA.P.013
- 7.3.3 Number of blades: 3
- 7.3.4 Diameter: 1.830 m (72 in) – No reduction is permitted
- 7.3.5 Sense of Rotation: Clockwise (pilot’s view)

8. Fluids

- 8.1 Fuel: AVGAS Grade 91/96 or 100 LL (ASTM D910) (see A.VI, note 3)
MOGAS EN 228 (E) (see A.VI, note 2)
Refer to doc. No. 2010/100 “P2010 Aircraft Flight Manual” for further details.

8.2 Oil:

Average Ambient Temperature	MIL-L-6082B or SAEJ1966 Spec. Mineral Grades	MIL-L-22851 or SAEJ1899 Spec. Ashless Dispersant Grades
All Temperatures	----	SAE15W50 or SAE20W-50
Above 80°F	SAE60	SAE60
Above 60°F	SAE50	SAE40 or SAE50
30°F to 90°F	SAE40	SAE40
0°F to 70°F	SAE30	SAE40, SAE30, SAE20W40
Below 10°F	SAE20	SAE30 or SAE20W30

Refer to Lycoming (L)IO-360-M1A “Operation and Installation Manual” and Lycoming (L)IO-390-C1B3 “Operation and Installation Manual” for list of alternative recommended commercial brands and types.

9. Fluid capacities:

- 9.1 Fuel:
- | | |
|----------|-----------------------------------|
| 2 Tanks: | 120 litres each (31.7 US gallons) |
| Total: | 240 litres (63.4 US gallons) |
| Usable: | 231 litres (61 US gallons) |
- 9.2.1 Oil:
- | | |
|--------|------------------------|
| Total: | 7.57 litres (8 US qts) |
|--------|------------------------|



	Minimum:	3.78 litres (4 US qts)	
9.2.2 Oil (see note 3):	Total:	6.62 litres (7 US qts)	
	Minimum:	3.78 litres (4 US qts)	
10. Air Speeds:	Never exceed speed V_{NE}		164 KCAS
	Maximum Structural Cruising Speed V_{NO}		130 KCAS
	Design Manoeuvring speed V_A		119 KCAS
			122 KCAS see A.VI, note 9
	Operating Manoeuvring speed V_O		119 KCAS
			122 KCAS see A.VI, note 9
	Maximum flaps extended speed V_{FE}		92 KCAS
			99 KCAS see A.VI, note 9
11. Maximum Operating Altitude:	12000 ft		
	14000 ft (see A.VI, note 7)		
12. Approved Operations Capability:	Day/Night-VFR, IFR ;		
	Refer to KOEL contained in the AFM, doc. No. 2010/100, Section 2.		
	Flight into expected or actual icing conditions is prohibited		
13. Maximum Masses:	Max Take-Off:	1160 kg (2557 lb)	
		1220 Kg (2690 lb) (see A.VI, note 9)	
	Max Landing:	1160 kg (2557 lb)	
		1220 Kg (2690 lb) (see A.VI, note 9)	
14. Centre of Gravity Range:	Forward Limit: 0.262 m (19% MAC) behind datum		
	Aft Limit: 0.440 m (32% MAC) behind datum		
	Mean Aerodynamic Chord is 1.378 m (54.2 in)		
14.1 Optional: (see A.VI, note 9)	Forward Limit: 0.220 m (16% MAC) behind datum up to 1100Kg		
	0.338 m (24,5% MAC) behind datum at 1220Kg		
	Straight line variations between indicated points.		
	Aft Limit: 0.440 m (32% MAC) behind datum for all weight		
	Mean Aerodynamic Chord is 1.378 m (54.2 in)		
15. Datum:	Vertical plane tangent to wing leading edge		
16. Control surface deflections:	Stabilator: $17^{\circ} \pm 2^{\circ}$ to pitch up / $6^{\circ} \pm 2^{\circ}$ to pitch down		
	Stabilator Trim Tab: $15 \pm 1^{\circ}$ downward / $3^{\circ} \pm 1^{\circ}$ upward		
	Stabilator Trim Tab: $6 \pm 1^{\circ}$ downward / $3^{\circ} \pm 1^{\circ}$ upward (see A.VI, note 4)		
	Aileron: $19^{\circ} \pm 2^{\circ}$ upward / $14^{\circ} \pm 2^{\circ}$ downward		
	Rudder: $25^{\circ} \pm 2^{\circ}$ left / $25^{\circ} \pm 2^{\circ}$ right		
	Rudder Trim Tab: $20^{\circ} \pm 2^{\circ}$ left / $20^{\circ} \pm 2^{\circ}$ right		
	Flaps: 0° Fully Retracted / $40^{\circ} \pm 1^{\circ}$ Fully Extended		
17. Levelling Means:	seat track supporting beams (see procedure in doc. No. 2010/100 "P2010 Aircraft Flight Manual", Section 6)		
18. Minimum Flight Crew:	1		
19. Maximum Passenger Seating Capacity:	3		



- | | |
|---------------------------------|--|
| 20. Baggage/Cargo Compartments: | Max Allowable Load: 40 kg (88 lb)
Location:1.56 m (61.41 in) from datum |
| 21. Wheels and Tyres: | Nose Wheel Tyre Size: 5.00-5, Type III
Main Wheel Tyre Size 6.00-6, Type III
For approved Types and rating see AMM, doc No. 2010/101 |
| 22. Serial Numbers Eligible: | See Note 5 |

A.IV. Operating and Service Instructions

- | | |
|---------------------------------|---|
| 1. Flight Manual: | Doc. No. 2010/100 "P2010 Aircraft Flight Manual" Last issue. |
| 2. Maintenance Manual: | Doc. No. 2010/101 "P2010 Aircraft Maintenance Manual" Last issue;
Airworthiness Limitations are reported in ATA chapter 4. |
| 3. Structural Repair Manual: | not available |
| 4. Weight and Balance Manual: | not available |
| 5. Illustrated Parts Catalogue: | Doc. No. 2010/102 "P2010 Illustrated Parts Catalogue" Last issue. |

A.V. Operational Suitability Data (OSD)

The Operational Suitability Data elements listed below are approved by the European Union Aviation Safety Agency under the EASA Type Certificate EASA.A.576 as per Commission Regulation (EU) 748/2012 as amended by Commission Regulation (EU) No 69/2014.

1. Master Minimum Equipment List (MMEL):
The MMEL is defined in the P2010 GEN.MMEL, Report n°2010/164, Revision 0 or later approved revisions.



A.VI. Notes:

- 1) When MOD 2010/002 (EASA approval 10052750) is installed
- 2) When MOD 2010/032 (EASA approval 10055692) is installed
- 3) When MOD 2010/078 (EASA approval 10065113) is installed
- 4) When MOD 2010/133 (EASA approval 10069356) is installed
- 5) Manufacturer's eligible serial numbers:
 - s/n 002 to subsequent for a/c manufactured by C.A. Tecnam S.P.A. under certificate EASA production certificate IT.21G.0032,
 - s/n CP-001 to subsequent for a/c manufactured by LUSY Co. LTD under certificate CAAC production certificate PC0034A-DBThe aircraft s/n CP-001 to subsequent can be delivered in China (including Hong Kong, Macao and Taiwan), Mongolia, North Korea & Pakistan and cannot be registered in Europe.
- 6) As per Manufacturer TCDS, propellers with designation having a "small" letter in the place of the brackets (for example "MTV-14-B-C-F/CF 195-30x") may be installed since it does not affect interchangeability. A capital letter in the place of the bracket (for example MTV-14-B-C-F/CF 195-30X) may not be installed according to propeller TCDS since it may affect interchangeability
- 7) When MOD 2010/194 (EASA approval 10073987) and MOD2010/078 (EASA approval 10065113) are installed
- 8) When MOD 2010/370 (EASA Approval No. 10089089) is installed. Available only when the Lycoming IO-390-C3B6 engine is installed
- 9) When MOD 2010/368 (EASA approval 10089247) is installed. Available only when the Lycoming IO-390-C3B6 engine is installed



SECTION B: P2010 TDI

B.I. General

- | | |
|---|-----------------------|
| 1. Type/Model/Variant | |
| 1.1 Type | P2010 |
| 1.2 Model: | P2010 TDI |
| 1.3 Variant | -- |
| 2. Airworthiness Category: | CS-23 Normal category |
| 3. Manufacturer: | see B.VI, note 1 |
| 4. EASA Type Certification
Application Date: | 29 April 2019 |
| 5. State of Design Authority | N/A |
| 6. State of Design Authority Type
Certificate Date | N/A |
| 7. EASA Type Certification Date | 08 October 2020 |

B.II. EASA Certification Basis

- | | |
|--|---|
| 1. Reference Date for
determining the applicable
requirements: | 29 April 2019 |
| 2. Airworthiness Requirements: | EASA CS-23 amdt.2 dated 28 September 2010
EASA CS-ACNS |
| 3. Special Conditions: | CRI B-52 (SC-B23.div-01 Human Factors – Integrated Avionic
System);
CRI F-58 (SC-F23.1353-02 Lithium Battery Installations)
CRI E-103 (para.1) Installation of the diesel engine TAE 125-02
CRI E-104 (SC-CS-23.1305- Fuel low level annunciation means)
- <i>see Explanatory Note TCDS EASA.A.576</i> |
| 4. Exemptions: | None |
| 5. Deviations: | None |
| 6. Equivalent Safety Findings: | CRI E-103 (para.3) Installation of the diesel engine TAE 125-02
- <i>see Explanatory Note TCDS EASA.A.576</i> |
| 7. Requirements elected to
comply: | EASA CS-23 amdt.4 para. 23.1306
EASA CS-23 amdt.4 para. 23.1308 |
| 8. Environmental Protection: | Refer to TCDSN EASA.A.576 |



B.III. Technical Characteristics and Operational Limitations

1. Type Design Definition: Document no. 2010/637 "Type Design Definition"
2. Description:
 - 2.1 Basic: Single-engine, variable pitch propeller, four seats, high wing aeroplane equipped with fixed tricycle landing gear, featuring composite, aluminium and steel construction.
3. Equipment: Equipment list, AFM, doc. No. 2010/552, Section 6
4. Dimensions:

Span	10.30 m (33.79 ft)
Length	7.91 m (25.95 ft)
Height	2.84 m (9.32 ft)
Wing Area	13.9 m ² (149.6 ft ²)
5. Engine:
 - 5.1 Model: Continental Engines: TAE 125-02-125
 - 5.2 Type Certificate: EASA Type Certificate No. EASA.E.055
 - 5.3 Limitations: Take-Off Power 125 kW (168HP) at 2300 RPM
Max continuous power 114 kW (153HP) at 2250 RPM
Other engine's limitations are listed in doc. No. 2010/552 "P2010 TDI Aircraft Flight Manual", Section 2
6. Load factors:

	Flap UP	Flap DOWN
Positive	+3.8 g	+2.0 g
Negative	-1.52 g	0.0 g
7. Propeller:
 - 7.1 Model: MT Propeller: MTV-6-R /190-69
 - 7.2 Type Certificate: EASA Type Certificate No. EASA.P.094
 - 7.3 Number of blades: 3
 - 7.4 Diameter: 1.900 m (75 in) – No reduction is permitted
 - 7.5 Sense of Rotation: Clockwise (pilot's view)
8. Fluids
 - 8.1 Fuel: JET A-1 (ASTM –D-1655)
Diesel (EN 590)
Refer to doc. No. 2010/552 "P2010 TDI Aircraft Flight Manual" for further details.
 - 8.2 Oil: Engine Aero Shell Oil Diesel Ultra, Shell Helix Ultra 5W30 or see applicable AFM, Section 2.
Gearbox Centurion Gearbox Oil N1, or see applicable AFM, Section 2
 - 8.3 Coolant Water / Cooler Protection
for more details see applicable AFM, Section 2
 - 8.4 Ice Protection Fluids: Liqui Moly "Diesel Fließ-Fit" or see applicable AFM, Section 2
9. Fluid capacities:
 - 9.1 Fuel: 2 Tanks: 120 litres each (31.7 US gallons)



	Total:	240 litres (63.4 US gallons)
	Usable:	231 litres (61 US gallons)
9.2. Oil:	Total:	6 litres (6.34 US qts)
	Minimum:	4.5 litres (4.75 US qts)
10. Air Speeds:	Never exceed speed V_{NE}	164 KCAS
	Maximum Structural Cruising Speed V_{NO}	130 KCAS
	Design Manoeuvring speed V_A	119 KCAS
		121 KCAS (see B.VI, note 2)
		122 KCAS (see B.VI, note 3)
	Operating Manoeuvring speed V_O	119 KCAS
		121 KCAS (see B.VI, note 2)
		122 KCAS (see B.VI, note 3)
	Maximum flaps extended speed V_{FE}	92 KCAS LND
		101 KCAS TO
		93 KCAS LND
		103 KCAS TO (see B.VI, note 2)
		94KCAS LND
		104 KCAS TO (see B.VI, note 3)
11. Maximum Operating Altitude:	18000 ft	
12. All-weather Operations Capability:	Day/Night-VFR, IFR; Refer to KOEL contained in the AFM, doc. No. 2010/552, Section 2. Flight into expected or actual icing conditions is prohibited	
13. Maximum Weights:	Max Take-Off:	1160 kg (2557 lb) 1200 Kg (2645 lb) (see B.VI, note 2) 1220 Kg (2690 lb) (see B.VI, note 3)
	Max Landing:	1160 kg (2557 lb) 1200 Kg (2645 lb) (see B.VI, note 2) 1220 Kg (2690 lb) (see B.VI, note 3)
14. Centre of Gravity Range:	Forward Limit: 0.275 m (19% MAC) behind datum up to 1000Kg 0.330 m (23% MAC) behind datum up to 1160kg, or - 0.344 m (24% MAC) behind datum up to 1200Kg (see B.VI, note 2), or - 0.351 m (24,5% MAC) behind datum up to 1220Kg (see B.VI, note 3). Straight line variations between indicated points. Aft Limit: 0.454 m (32% MAC) behind datum for all weight Mean Aerodynamic Chord is 1.378 m (54.2 in)	
15. Datum:	Vertical plane tangent to wing leading edge	



16. Control surface deflections: Stabilator TE: $17^{\circ} \pm 2^{\circ}$ upward/ $6^{\circ} \pm 2^{\circ}$ downward
Stabilator Trim Tab TE (Stabilator 0°): $8^{\circ} \pm 2^{\circ}$ downward/ $2^{\circ} \pm 2^{\circ}$ upward
Aileron TE: $19^{\circ} \pm 2^{\circ}$ upward/ $14^{\circ} \pm 2^{\circ}$ downward
Rudder TE: $25^{\circ} \pm 2^{\circ}$ left/ $25^{\circ} \pm 2^{\circ}$ right
Rudder Trim Tab TE: $20^{\circ} \pm 2^{\circ}$ left/ $20^{\circ} \pm 2^{\circ}$ right
Flaps TE: 0° Fully Retracted/ $40^{\circ} \pm 1^{\circ}$ Fully Extended
17. Levelling Means: seat track supporting beams (see procedure in doc. No. 2010/552
"P2010 TDI Aircraft Flight Manual", Section 6)
18. Minimum Flight Crew: 1
19. Maximum Passenger Seating Capacity: 3
20. Baggage/Cargo Compartments: Max Allowable Load: 40 kg (88 lb)
Location: 1.56 m (61.41 in) from datum
21. Wheels and Tyres: Nose Wheel Tyre Size: 5.00-5, Type III
Main Wheel Tyre Size 6.25-6, Type III
For approved Types and rating see AMM, doc No. 2010/553
22. Serial Numbers Eligible: See B.VI, Note 1



B.IV. Operating and Service Instructions

- | | |
|---------------------------------|---|
| 1. Flight Manual: | Doc. No. 2010/552 "P2010 TDI Aircraft Flight Manual" Last issue. |
| 2. Maintenance Manual: | Doc. No. 2010/553 "P2010 TDI Aircraft Maintenance Manual" Last issue;
Airworthiness Limitations are reported in ATA chapter 4. |
| 3. Structural Repair Manual: | not available |
| 4. Weight and Balance Manual: | not available |
| 5. Illustrated Parts Catalogue: | Doc. No. 2010/638 "P2010 TDI Illustrated Parts Catalogue" Last issue. |

B.V. Operational Suitability Data (OSD)

The Operational Suitability Data elements listed below are approved by the European Union Aviation Safety Agency under the EASA Type Certificate EASA.A.576 as per Commission Regulation (EU) 748/2012 as amended by Commission Regulation (EU) No 69/2014.

1. Master Minimum Equipment List (MMEL)
The MMEL is defined in the P2010 GEN.MMEL, Report n°2010/164, Revision 0 or later approved revisions.



B.VI. Notes:

1) Manufacturer's eligible serial numbers:

S/N 100 to subsequent (when MOD2010/162 is installed - EASA approval 10074522) for A/C
manufactured by C.A. Tecnam S.P.A. under certificate EASA production certificate IT.21G.0032,

2) When MOD 2010/207 (EASA approval 10076578) is installed

3) When MOD 2010/269 (EASA approval 10081499) is installed



SECTION ADMINISTRATIVE

I. Acronyms & Abbreviations

AFM – Aircraft Flight Manual
AMM – Aircraft Maintenance Manual
ASTM – American Society for Testing and Materials
CRI – Certification Review Item
CS – Certification Specification
EASA – European Union Aviation Safety Agency
ICAO – International Civil Aviation Organization
IPC – Illustrated Part Catalogue
KCAS – Knots Calibrated Air Speed
KOEL – Kind of Operations Equipment List
MAC – Mean Aerodynamic Chord
MLW – Maximum Landing Weight
MTOW – Maximum Take-Off Weight
MZFW – Maximum Zero Fuel Weight
TC – Type Certificate
TCDS – Type Certificate Data Sheet
VFR – Visual Flight Rules
IFR – Instrumental Flight Rules
TE – Trailing Edge

II. Type Certificate Holder Record

TC Holder	Period
Costruzioni Aeronautiche TECNAM S.r.l. Via Tasso, 478 80127 Napoli, ITALIA	From 26 th September 2014 until 26 th August 2019
Costruzioni Aeronautiche TECNAM S.p.A. Via Salvo D'acquisto 62 80042 Boscotrecase, Napoli ITALIA	Effective



III. Change Record

Issue	Date	Changes	TC Issue Date
Issue 01	26 Sept 2014	Initial Issue	26 Sept 2014
Issue 02	05 May 2015	MT Variable Pitch Propeller Added	
Issue 03	16 Dec 2015	Update to include changes: MOD2010/001 "GFC 700 autopilot" (EASA approval 10055187), MOD2010/003 "Alternative avionics configuration" (EASA approval 10053996), MOD2010/032 Automobile fuel (EASA approval 10055692)	
Issue 04	22 Dec 2016	Introduction of OSD MMEL. CRI F-102 (and corresponding note 3) has been removed since it is not a special condition	
Issue 05	29 March 2018	Amended to include change MOD2010/078 (EASA approval 10065113)	
Issue 06	25 March 2019	Amended to include change MOD2010/133 (EASA approval 10069356), remove typos and update company business registration.	
Issue 07	23 May 2019	Added Chinese manufacturer, updated eligible s/n and Company address	
Issue 08	20 Dec 2019	Updated propeller designation (field A.III (7.2 and 7.3). Added note 6	
Issue 09	07 Aug 2020	Amended to remove typo and include change MOD2010/194 (EASA approval 10073987)	
Issue 10	08 Oct 2020	Amended to included P2010 TDI model (MOD2010/162 –EASA approval 10074522)	08 October 2020
Issue 11	31 May 2021	Amended to include change MOD2010/207 (EASA approval 10076578) and remove typo in notes reference	
Issue 12	23 March 2023	Amended to include change MOD2010/269 (EASA approval 10081499), included reference to TCDSN and improved control surface deflection information on P2010TDI	
Issue 13	11 Feb 2026	§A.III.4 Dimensions: height amended, ref. design change 2010/370. §§ A.III.10 Airspeeds, A.III.13 Maximum Masses, A.III.14 Centre of Gravity range: amended to include values at MTOW 1220Kg (ref. 2010/368). § A.VI Notes: note 8 (MOD2010/370) and note 9 (MOD2010/368) added. § B.III.21 Wheels and Tyres: editorial correction.	

