This annex to TCDS EASA.A.037 was created to publish selected special conditions / deviations / equivalent safety findings that are part of the applicable certification basis:

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SPECIAL CONDITION	F-23 SC: Non-rechargeable Lithium Battery Installations
APPLICABILITY:	F28 all marks
REQUIREMENTS:	CS 25.601, 25.863, 25.869, 25.1301, 25.1309, 25.1353(c), 25.1529, 25.1360 (b)
ADVISORY MATERIAL:	N/A

BACKGROUND

The current requirements governing the installation of batteries in Large Aeroplanes are covered under (CS) 25.1353(c). Requirements from (CS) 25.1353(c) are essentially unchanged from initial JAR code and do not adequately address several failure, operational, and maintenance characteristics of Li-Batteries that could affect safety and reliability of those battery installations.

Recent events involving a non-rechargeable lithium battery system in emergency locator transmitter (ELT) installations have identified unanticipated failure modes associated with non-rechargeable lithium battery installations. These may include over-discharging, cell imbalance, external short circuit, internal short circuit, and flammability of cell components among other possibilities.

Non-rechargeable lithium batteries, in design and operation, are different than nickel-cadmium and leadacid non-rechargeable batteries. While the non-rechargeable lithium battery concept is not itself novel, higher energy levels are being introduced into aircraft systems by adopting new chemical composition in various battery cell sizes and construction. Interconnection of these cells in battery packs introduces failure modes that require unique design considerations, such as provisions for thermal management.

In addition to the ELT which was the subject of the referenced ADs, known uses of non-rechargeable lithium batteries and battery systems on aircraft include:

- Flight deck and avionics systems such as displays, global positioning systems, cockpit voice recorders, flight data recorders, underwater-locator-beacons, navigation computers, integrated avionics computers, satellite network/communication systems, communication management units, and remote monitor electronic line replaceable units (LRU);
- Cabin safety, entertainment and communications equipment including life rafts, escape slides, seat belt air bags, cabin management systems, Ethernet switches, routers and media servers, wireless systems, internet/in-flight entertainment systems, satellite televisions, remotes and handsets;
- Systems in cargo areas including door controls, sensors, video surveillance equipment and security systems.

Some known potential hazards and failure modes associated with non-rechargeable lithium batteries and battery systems are described below.

Internal Failures

In general, lithium batteries are significantly more susceptible to internal failures that can result in self-sustaining increases in temperature and pressure (i.e., thermal runaway) than their nickel-cadmium or lead-acid counterparts. The metallic lithium can ignite, resulting in a self-sustaining fire and/or explosion.

Fast-discharging

An imbalanced discharge of one cell of a multi-cell non-rechargeable lithium battery system may create an overheating condition that results in an uncontrollable venting condition, which in turn could lead to a thermal event and/or an explosion.

Flammability

Unlike nickel-cadmium and lead-acid batteries, lithium batteries and battery systems use higher energy and current in an electrochemical system that can be configured to maximize energy storage of lithium and use liquid electrolytes that can be extremely flammable. The electrolyte, as well as the electrodes, can serve as a source of fuel for an external fire if there is a breach of the battery case.

The intent of this CRI is to ensure that these non-rechargeable Lithium battery installations are not unsafe, to an extent necessary to support issuance of an airworthiness certificate.

SPECIAL CONDITION

In lieu of the requirements of CS 25.1353(c) (1) through (c)(4), non-rechargeable Lithium batteries and battery installations must comply with the following special conditions:

- 1. Be designed so that safe cell temperatures and pressures are maintained under all foreseeable operating conditions to preclude fire and explosion.
- 2. Be designed to preclude the occurrence of self-sustaining, uncontrolled increases in temperature or pressure.
- 3. Not emit explosive or toxic gases in normal operation, or as a result of its failure, that may accumulate in hazardous quantities within the airplane.
- 4. Must meet the requirements of CS 25.863(a) through (d).
- 5. Not damage surrounding structure or adjacent systems, equipment or electrical wiring of the airplane from corrosive fluids or gases that may escape and that may cause a major or more severe failure condition.
- 6. Have provisions to prevent any hazardous effect on airplane structure or essential systems caused by the maximum amount of heat it can generate due to any failure of it or its individual cells.
- 7. Have a means to detect its failure and alert the flight crew in case its failure affects safe operation of the aircraft.
- 8. Have a means for the flight crew or maintenance personnel to determine the battery charge state if its function is required for safe operation of the airplane.

Note 1: A battery system consists of the battery and any protective, monitoring and alerting circuitry or hardware inside or outside of the battery. It also includes vents (where necessary) and packaging. For the purpose of this special condition, a battery and battery system are referred to as a battery.

Note 2: These special conditions apply to all non-rechargeable lithium battery installations in lieu of 25.1353(c)(1) through (c)(4). Section 25.1353(c)(1) through (c)(4) will remain in effect for other battery installations.

Note 3: For Very Small Non-rechargeable Lithium Batteries (equal or less than 2 Watt-hour of energy), an acceptable MoC with this Special Conditions is showing these batteries compliant with Underwriters Laboratories (UL) 1642.

Note 4: For the purpose of SCs 7 and 8, "safe operation of the airplane" is defined as continued safe flight and landing following failures or other non-normal conditions. The following are examples of devices with batteries that are not required for continued safe flight and landing of the airplane: emergency locator transmitters, underwater locator beacons, seat belt air bag initiators and flashlights. A backup flight instrument with a non-rechargeable lithium battery is an example that would be required for safe operation of the airplane.

- END -

ACRONYMS AND ABBREVIATIONS

ADS Automatic Dependent Surveillance

AC Advisory Circular
ACJ Advisory Circular Joint
AFM Airplane Flight Manual

AMC Acceptable Means of Compliance

AMJ Advisory Material Joint
AMM Aircraft Maintenance Manual

AOA Angle Of Attack
APU Auxiliary Power Unit

ARAC Aviation Rulemaking Advisory Committee
ASPSU Auxiliary Standby Power Supply Unit

ATN Air Traffic Network
ATS Air Traffic Service

AVM Airborne Vibration Monitor
AWO All Weathers Operations
CC Cargo Compartment

CCRC Cabin Crew Rest Compartments
CFD Computational Fluid Dynamics

CLS Cargo Loading System

CLTA Conventional Large Transport AircraftCMR Certification Maintenance Requirements

CFR Code of Federal RegulationsCFRP Carbon Fiber Reinforced Plastic

CPDLC Controller-Pilot Data Link Communications

CRC Crew Rest Compartments
CRI Certification Review Item

CRT Cathode Ray Tube
CVR Cockpit Voice Recorder

DEV Deviation

DFZ Designated Fire Zones

D-FIS Data Link Flight Information Services

DOP Delegation Of Power

DOT Department of Transportation

DH Decision Height

DRI Dynamic Response IndexDSG Design Service GoalEAS Equivalent Airspeed

ECAM Electronic Centralized Aircraft Monitoring

EFC Engine Electronic Control
 EFCS Electrical Flight Control System
 ELOS Equivalent Level of Safety
 EPF Emergency Passage Feature
 EPS Executive Power System

EPSUs Emergency Power Supply Units

ESF Equivalent Safety Finding

ETSO European Technical Standard Order

FCRC Flight Crew Rest Compartments

FDR Flight Data Recorder

FMJ RN Full Metal Jacket, Round Nose **FQIS** Fuel Quantity Indicating Systems

FWS Flight Warning System
GFI Ground Fault Interruption
GVT Ground Vibration Test
HIC Head Injury Criterion

HIRF High Intensity Radiated Fields

HP High Pressure

HWG Harmonisation Working Group

ICA Instructions for Continued Airworthiness ICAO International Civil Aviation Organisation

ID Inside Diameter

IDF Imbalance Design Fraction
IFE In-Flight Entertainment
IFSD In Flight Shut Down
ILD Inertia Locking Device
IM Interpretative Material

IMC Instrument Meteorological Conditions

IP Intermediate Pressure

ISPSS In-Seat Power Supply System

JHP Jacketed Hollow Point

JTSO Join Technical Service Order

LCD Liquid Crystal Display

LRS Lower Deck Cargo Compartment Limit of Reasonable Survivability

LP Low Pressure

MMEL Master Minimum Equipment List

MOC Means Of Compliance

NPA Notice of Proposal AmendmentNPRM Notice of Proposed RulemakingNTSB National Transportation Safety Board

OSD Operational Suitability Data
PBE Protective Breathing Equipment

PDU Power Drive Unit

PED Portable Electronic Device

PPHWG Power Plant Harmonisation Working Group

RMS Root Mean Square

RR Rolls Royce

RTCA Radio Technical Commission for Aeronautics

RVR Runway Visual Range SC Special Condition TAS True Airspeed

TCDS Type Certificate Data Sheet
TSO Technical Service Order
TT&L Taxi, Tack off and Landing

TWA Trans World Airline

VLTA Very Large Transport Aeroplane

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