

VTOL Design Loads and Interaction of Systems and Structures

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Design Loads VTOL MoC

- → General
- → Design Load Requirements:
 - → VTOL.2200 Structural Design Envelope
 - → VTOL.2215 Flight loads conditions
 - \rightarrow VTOL.2205 Interaction of systems and structures
 - → VTOL.2220 Ground and water loads
- → Summary





Different safety objectives:

Different configurations and modes:

Flight

Envelope



Flight control system interaction with flight envelope:

Flight Control System



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VTOL.2200 Structural Design Envelope

- → Proposed MOC published 25 May 2020
- → 15 comments reviewed and MOC under revision



MOC PHASE 1



VTOL.2200 Design Airspeeds



All configurations and modes should be considered for design airspeed(s) definition: more than one airspeed definition set may be necessary



MOC PHASE 2

VTOL.2215(a) Flight Load Conditions

- → Proposed MOC published 25 May 2020
- → 19 comments reviewed and MOC under revision



Symmetrical pull-up and recovery Max positive load factor Max associated pitch accelerations



Symmetrical pushover and recovery Max negative load factor Max associated pitch accelerations



1g Symmetric Flight



Rolling Flight Conditions 2/3 max positive load factor Max associated pitch and roll accelerations



Design Airspeeds / Flight Loads

EASA



All critical speeds and configurations should be considered for each flight manoeuvre, up to the aircraft maximum defined: VD(s), VH(s) or VNE(s)

VTOL.2205 Interaction of systems and structures

System failures:





VTOL.2205 Interaction of systems and structures

ADAPTED

MOC VTOL.2205

any structure the loading of which may

be changed by failure(s) of the system

 \rightarrow MOC based on CS 25 Appendix K

CS 25 Appendix K structure whose failure could prevent continued safe flight and landing

→ Scenarios to consider:

System	System in failure condition		Failure	Dispatch
fully	At time of occurrence	Continuation of the flight	indication	with known
operable	Static Strength ⁽¹⁾	Static Strength ⁽¹⁾		failure
	Residual Strength	Residual Strength		conditions
Nominal	Vibrations	Vibrations	Detectability	Limitations
condition	Flutter (if failure causes	Flutter ⁽¹⁾	· ·	may be
	velocity increase)	Fatigue & Damage Tolerance		established

⁽¹⁾ For determination of **Safety Factor** and **Flutter Speed**, the probability will be consistent with the safety objective defined in SC VTOL for Category Enhanced and Category Basic (no. passengers)

Reference: MOC VTOL.2510 Equipment, systems, and installations





VTOL.2220 Ground and Water Loads

Ground conditions:

- Proposed MOC published 25 May 2020
- ✓ 3 comments reviewed and MoC under revision



Water conditions:

• Seaplane and amphibian water landing loads to be addressed in Phase 3 MOC



Summary

- → Complexity in VTOL design load definition: different safety objectives, configurations and modes, and complex flight control systems
- → Interaction of systems and structures needs comprehensive analysis to consider all failures that could influence loading and flutter
- → Many valuable comments received during Phase 1 consultation will lead to improvement and simplification of the proposed design load MoCs
- → Further design load MoCs will be released for public consultation in Phases 2 and 3





Thank you for your attention

Feel free to submit your questions on our live event platform.....





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