TYPE CERTIFICATE
DATA SHEET

No. EASA.IM.R.003

for
S-64

Type Certificate Holder
Erickson Incorporated, DBA Erickson Air-Crane
3100 Willow Springs Road
Central Point, Oregon, 97502-0010
U.S.A.

For Models:  S-64F
            S-64E
# TABLE OF CONTENTS

SECTION 1: S-64F .................................................................................................................. 3
   I. General......................................................................................................................... 3
   II. Certification Basis ...................................................................................................... 3
   III. Technical Characteristics and Operational Limitations ........................................... 3
   IV. Operating and Service Instructions ........................................................................... 6
   V. Notes .......................................................................................................................... 6

SECTION 2: S-64E .................................................................................................................. 7
   I. General......................................................................................................................... 7
   II. Certification Basis ...................................................................................................... 7
   III. Technical Characteristics and Operational Limitations ........................................... 7
   IV. Operating and Service Instructions ........................................................................... 10
   V. Notes .......................................................................................................................... 10

SECTION 3: OPERATIONAL SUITABILITY DATA (OSD) ......................................................... 11
   I. OSD Certification Basis ............................................................................................... 11
   II. OSD Elements .......................................................................................................... 11

SECTION: ADMINISTRATIVE ................................................................................................. 12
   I. Acronyms and Abbreviations ..................................................................................... 12
   II. Type Certificate Holder Record ................................................................................. 12
   III. Change Record ......................................................................................................... 12
SECTION 1: S-64F

I. General

1. Type / Model
   1.1 Type
   S-64
   1.2 Model
   S-64F

2. Airworthiness Category
   Large Rotorcraft

3. Manufacturer
   Erickson Incorporated, DBA Erickson Air-Crane
   3100 Willow Springs Road,
   Central Point, Oregon 97502-0010, U.S.A.;
   See also Note 2.

4. Type Certification Application Date
   to FAA: 2 April 1969
   to ENAC: 15 February 1999

5. State of Design Authority
   Federal Aviation Administration (USA)

6. Type Certificate Date by FAA
   25 November 1970

7. Type Certificate n° by FAA
   H6EA

8. Type Certificate Data Sheet n° by FAA
   H6EA

9. EASA Type Certification Date
   4 August 2004

II. Certification Basis

1. Reference Date for determining the applicable requirements
   2 April 1969

2. Airworthiness Requirements
   FAR 29, dated 1 February 1965 including Amdts. 29-1 and 29-2, except FAR 29.855 (d).
   Note: The FAA reference date of application as of 2 April 1969 was accepted and retained as reference date for EASA Certification Basis. Since the JAR Requirements were not in existence at the reference date, the FAA Certification Basis is retained as EASA Certification Basis.

3. Special Conditions
   29-014-SC

4. Exemptions
   none

5. Deviations
   none

6. Equivalent Safety Findings
   none

7. Environmental Protection Requirements
   7.1 Noise Requirements
   n/a, the S-64F is designed and intended to be operated exclusively for external load carrying purpose

   7.2 Emission Requirements
   ICAO Annex 16, Volume 11

8. Operational Suitability Data (OSD)
   See SECTION III below.

III. Technical Characteristics and Operational Limitations

1. Type Design Definition
   EAC drawing, 6401-10015 Rev. D, and subsequent EASA approved revisions.
   Note 1: Installation of Elastomeric Engine Mounts
   P/N EA643021-109 is required
   Note 2: Type design change 6450-10522 (Shoulder Harness on Crewman Seat), 6455-61337 (Spare AC fuses for Generator), 6430-10616 and 6430-63050 (Environmental
2. Description
Transport Rotorcraft designed as ‘Industrial Flying Crane Helicopter’ and primarily intended to carry cargo in external load operations up to 11 340 kg by means of hydraulic hoist or cargo hook. With EAC STC SR00004SE installed also certified for firefighting.

- Main rotor: Six (6) blades
- Tail rotor: Four (4) blades
- Fuselage: Traditional Aluminium fuselage structure
- Landing gear: Fixed tricycle landing gear
- Powerplant: Two (2) turbine engines with APU

3. Equipment
Basic equipment must be installed and operational prior to registration of the helicopter. Refer to Equipment list in approved RFM

4. Dimensions

4.1 Fuselage
- Length: 27.23 m (88 ft 6 in)
- Width: 6.71 m (21 ft 10 in)
- Height: 7.82 m (25 ft 5 in)

4.2 Main Rotor
- Diameter: 21.95 m (72 ft)

4.3 Tail Rotor
- Diameter: 4.88 m (16 ft)

5. Engine
5.1 Model
Erickson Incorporated (former: Pratt & Whitney)
2 x Model JFTD12A-5A
(with Hamilton Standard Fuel Control JFCS6-6)

5.2 Type Certificate
- FAA TC/TCDS n°: E15EA
- ENAC TC/TCDS n°: MO-108
- EASA TC/TCDS n°: EASA.IM.E.106

5.3 Limitations
5.3.1 Installed Engine Limitations

<table>
<thead>
<tr>
<th></th>
<th>PWR turbine rpm (%N2)</th>
<th>Gas generator rpm (%N1)</th>
<th>PWR turbine inlet temperature T5 [°C]</th>
<th>PWR [shp]</th>
</tr>
</thead>
<tbody>
<tr>
<td>AEO-TOP (5 min)</td>
<td>9 500 (105)</td>
<td>16 700 (104.2)</td>
<td>720</td>
<td>4 800</td>
</tr>
<tr>
<td>AEO-MCP</td>
<td>9 500 (105)</td>
<td>16 700 (104.2)</td>
<td>720</td>
<td>4 430</td>
</tr>
<tr>
<td>OEI (30 min)</td>
<td>9 500 (105)</td>
<td>16 700 (104.2)</td>
<td>675</td>
<td>4 800</td>
</tr>
<tr>
<td>Max. allowable overspeed</td>
<td>10 350 (114)</td>
<td>16 700 (104.2)</td>
<td>not defined</td>
<td>- - -</td>
</tr>
<tr>
<td>Acceleration limit (2 min)</td>
<td>not defined</td>
<td>not defined</td>
<td>720</td>
<td>- - -</td>
</tr>
<tr>
<td>Starting limit (2 sec)</td>
<td>not defined</td>
<td>not defined</td>
<td>525</td>
<td>- - -</td>
</tr>
</tbody>
</table>

5.3.2 Other Engine and Transmission Torque Limits
Refer to approved RFM

6. Fluids
For detailed information, see Section 1 of the Rotorcraft Flight Manual

6.1 Fuel
Jet A or Jet A-1 or Jet B or JP-4 or JP-5 or JP-8+100 (conforming to Pratt & Whitney Aircraft SB 2016)

6.2 Oil
- Engine: as per P&W S.B. 238
- APU: refer to approved RFM
6.3 Additives
n/a

7. Fluid capacities

7.1 Fuel
Fuel tank capacity: total 5 133 litres (1 356 US gal)
Usable fuel:
- 1 719 litres (454 US gal) at +280.8
- 1 719 litres (454 US gal) at +397.3
- 1 696 litres (448 US gal) at +461.3
Unusable fuel: total 26 lb
- 10 lb at +290.0
- 9 lb at +370.0
- 7 lb at +461.0

7.2 Oil
Engine: total 9.84 litres (2.6 US gal)
- 2 x 4.92 litres (1.3 US gal) at +234.0
Undrainable oil: 5 lb at +234.0
APU: not recorded

7.3 Coolant System Capacity
n/a

8. Air Speed Limitations
V_{NE}: 104 kt (120 mph) at 21 319 kg (47 000 lb)
Refer to approved RFM for other limitations.

9. Rotor Speed Limitations
Power on:
Maximum 104% N_{R} (193 rpm)
Minimum 100% N_{R} (185 rpm)
Power off:
Maximum 110% N_{R} (204 rpm)
Minimum 95% N_{R} (176 rpm)

10. Maximum Operating Altitude and Temperature
10.1 Altitude
16 000 ft DA (4 877 m)
10.2 Temperature
not recorded

11. Operating Limitations
- VFR day
- Logging operations are not allowed

12. Maximum Mass
TKOF: 21 319 kg (47 000 lb)
Refer to approved RFM for variations of maximum allowable weight with temperature and altitude.

13. Centre of Gravity Range
Refer to approved RFM

14. Datum
Longitudinal:
the datum plane (STA 0) is located 8 534 mm (336 in) forward of main rotor centroid

15. Levelling Means
Plumb line from top level plate inside cockpit aft door

16. Minimum Flight Crew
two (2), pilot and co-pilot

17. Maximum Occupant Seating Capacity
5
2 at +94.0 in, 1 at +108.5 in, 1 at +127.0 in,
1 at +130.0 in

18. Emergency Exit
not recorded

19. Maximum Baggage/Cargo Loads
227 kg (500 lb)
Two baggage compartments with max. floor loading of 21.1 kg/cm² (300 lb/ft²) and a total allowable load of 113.5 kg (250 lb) in each compartment.

20. Rotor Blade Control Movement
For rigging information refer to Maintenance Manual

21. Auxiliary Power Unit (APU)
SOLAR T-62T-16A1, or T-62T-16A2

22. Life-limited Parts
See Service Bulletin: S-64F General-1
23. Wheels and Tyres

Tyres: not recorded
Wheels: not recorded

IV. Operating and Service Instructions

1. Flight Manual

S-64F Rotorcraft Flight Manual Publication SA4047-5


Publication No. EAC006
The Airworthiness Limitations are listed in the S-64F Service Bulletins:
S-64F General-1, S-64F General-2 and S-64F General-3

3. Service Letters and Service Bulletins

As published by Sikorsky Aircraft, Erickson Air-Crane Co.,
Erickson Air-Crane Co., L.L.C., Erickson Air-Crane Inc., DBA Erickson Air-Crane and Erickson Inc., DBA Erickson Air-Crane.


Refer to approved RFM

5. Illustrated Parts Catalogue

n/a

6. Miscellaneous Manuals

not recorded

7. Required Equipment

The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification.

In addition, the following item of equipment is required:

V. Notes

1. Manufacturer's eligible serial numbers:

64067, 64078, 64080, 64081, 64084 until 64086, 64088 until 64091, 64093, 64095, 64097, 64098, 64F5001, and subsequent.

2. Type Certificate holder record note as per FAA TCDS H6EA, Revision 15, dated 17 March 2015:

- ‘Erickson Air-Crane Incorporated, DBA Erickson Air-Crane’ transferred TC H6EA to ‘Erickson Incorporated, DBA Erickson Air-Crane’ on 13 August 2014;

- ‘Erickson Air-Crane Co., L.L.C.’ transferred TC H6EA to ‘Erickson Air-Crane Incorporated, DBA Erickson Air-Crane’ on 14 February 2001;

- ‘Erickson Air-Crane Co.’ transferred TC H6EA to ‘Erickson Air-Crane Co., L.L.C.’ on 22 August 1997;


* * *
SECTION 2: S-64E

I. General

1. Type / Model
   1.1 Type S-64
   1.2 Model S-64E

2. Airworthiness Category
   Large Rotorcraft

3. Type Certificate Holder
   Erickson Incorporated, DBA Erickson Air-Crane
   3100 Willow Springs Road,
   Central Point, Oregon 97502-0010, U.S.A.;
   See also Note 2.

4. Type Certification Application Date to FAA: 27 November 1967
   to EASA: 4 August 2020

5. State of Design Authority
   Federal Aviation Administration (USA)

6. Type Certificate Date by FAA
   21 August 1969

7. Type Certificate n° by FAA
   H6EA

8. Type Certificate Data Sheet n° by FAA
   H6EA

9. EASA Type Certification Date
   16 December 2021

II. Certification Basis

1. Reference Date for determining the applicable requirements
   27 November 1967

2. Airworthiness Requirements
   FAR 29, dated 1 February 1965 including Amdts. 29-1 and
   29-2, except FAR 29.855 (d).

   Note: The FAA reference date of application as of 27
   November 1967 was accepted and retained as reference
date for EASA Certification Basis. Since the JAR
Requirements were not in existence at the reference
date, the FAA Certification Basis is retained as EASA
Certification Basis.

3. Special Conditions
   29-014-SC

4. Exemptions
   none

5. Deviations
   none

6. Equivalent Safety Findings
   none

7. Environmental Protection Requirements
   7.1 Noise Requirements
   n/a,
   the S-64F is designed and intended to be operated
   exclusively for external load carrying purpose

   7.2 Emission Requirements
   ICAO Annex 16, Volume II

8. Operational Suitability Data (OSD)
   See SECTION 3 below.

III. Technical Characteristics and Operational Limitations

1. Type Design Definition
   EAC drawing. 6401-10021 Rev. B, and subsequent EASA
   approved revisions, and FAA Supplemental Type
   Certificates SR00004SE, SR00026SE, SR00502SE,
   SR00815SE, SH606NW, SH1618NM, SH5102NM, and
   subsequent EASA approved revisions.
Note 1: Main Rotor Inner Bracket P/N S1510-21332-043 is required.
Note 2: Main Rotor Blades P/N 6415-20201-048, -049, -050 or 051 are required.

2. Description
Transport Rotorcraft designed as ‘Industrial Flying Crane Helicopter’ and primarily intended to carry cargo in external load operations up to 9 072 kg by means of hydraulic hoist or cargo hook. With EAC STC SR00004SE installed also certified for firefighting.

Main rotor: Six (6) blades
Tail rotor: Four (4) blades
Fuselage: Traditional Aluminium fuselage structure
Landing gear: Fixed tricycle landing gear
Powerplant: Two (2) turbine engines with APU

3. Equipment
Basic equipment must be installed and operational prior to registration of the helicopter.
Refer to Equipment list in approved RFM

4. Dimensions
4.1 Fuselage
Length: 27.23 m (88 ft 6 in)
Width: 6.71 m (21 ft 10 in)
Height: 7.82 m (25 ft 5 in)

4.2 Main Rotor
Diameter: 21.95 m (72 ft)

4.3 Tail Rotor
Diameter: 4.88 m (16 ft)

5. Engine
5.1 Model
Erickson Incorporated (former: Pratt & Whitney)
2 x Model JFTD12A-4A
(with Hamilton Standard Fuel Control JFC56-4)

5.2 Type Certificate
FAA TC/TCDS n°: E15EA
ENAC TC/TCDS n°: MO-108
EASA TC/TCDS n°: EASA.IM.E.106

5.3 Limitations
5.3.1 Installed Engine Limitations

<table>
<thead>
<tr>
<th></th>
<th>PWR turbine [rpm (%N2)]</th>
<th>Gas generator [rpm (%N1)]</th>
<th>PWR turbine inlet temperature T5 [°C]</th>
<th>PWR [shp]</th>
</tr>
</thead>
<tbody>
<tr>
<td>AEO-TOP (5 min)</td>
<td>9 500 (105)</td>
<td>16 700 (104.2)</td>
<td>688</td>
<td>4 500</td>
</tr>
<tr>
<td>AEO-MCP</td>
<td>9 500 (105)</td>
<td>16 700 (104.2)</td>
<td>655</td>
<td>4 000</td>
</tr>
<tr>
<td>OEI (30 min)</td>
<td>9 500 (105)</td>
<td>16 700 (104.2)</td>
<td>675</td>
<td>4 500</td>
</tr>
<tr>
<td>Max. allowable overspeed</td>
<td>10 350 (114)</td>
<td>16 700 (104.2)</td>
<td>not defined</td>
<td>- - -</td>
</tr>
<tr>
<td>Acceleration limit (2 min)</td>
<td>not defined</td>
<td>not defined</td>
<td>688</td>
<td>- - -</td>
</tr>
</tbody>
</table>

Notes:
- Sea level static, standard day conditions
- Take-off and maximum continuous horsepower ratings are normally obtained at a power turbine speed of 9 000 rpm (100%N2).
- Total power for two-engine operation is limited to 6 600 shp for take-off, and 5 400 shp maximum continuous.

5.3.2 Other Engine and Transmission Torque Limits
Refer to approved RFM
6. Fluids

6.1 Fuel

For detailed information, see Section 1 of the Rotorcraft Flight Manual

6.2 Oil

Jet A or Jet A-1 or Jet B or JP-4 or JP-5 or JP-8+100 (conforming to Pratt & Whitney Aircraft SB 2016)

6.3 Additives

Engine: as per P&W S.B. 238
APU: refer to approved RFM

n/a

7. Fluid capacities

7.1 Fuel

Fuel tank capacity: total 5 133 litres (1 356 US gal)
Usable fuel:
- 1 719 litres (454 US gal) at +280.8
- 1 719 litres (454 US gal) at +397.3
- 1 696 litres (448 US gal) at +461.3

Unusable fuel: total 26 lb
- 10 lb at +290.0
- 9 lb at +370.0
- 7 lb at +461.0

7.2 Oil

Engine:
- total 9.84 litres (2.6 US gal)
- 2 x 4.92 litres (1.3 US gal) at +234.0

Undrainable oil: 5 lb at +234.0
APU: not recorded

n/a

7.3 Coolant System Capacity

n/a

8. Air Speed Limitations

$V_{NE}: 99$ kt (114 mph) at 19 051 kg (42 000 lb)
Refer to approved RFM for other limitations.

9. Rotor Speed Limitations

Power on:
- Maximum $104\% N_R$ (193 rpm)
- Minimum $100\% N_R$ (185 rpm)

Power off:
- Maximum $110\% N_R$ (204 rpm)
- Minimum $89\% N_R$ (165 rpm)

10. Maximum Operating Altitude and Temperature

10.1 Altitude

14 000 ft DA (4 267 m)

10.2 Temperature

not recorded

11. Operating Limitations

- VFR day
- Logging operations are not allowed

12. Maximum Mass

TKOF: 19 051 kg (42 000 lb)
Refer to approved RFM for variations of maximum allowable weight with temperature and altitude.

13. Centre of Gravity Range

Refer to approved RFM

14. Datum

Longitudinal:
the datum plane (STA 0) is located 8 534 mm (336 in) forward of main rotor centroid

15. Levelling Means

Plumb line from top level plate inside cockpit aft door

16. Minimum Flight Crew

two (2), pilot and co-pilot

17. Maximum Occupant Seating Capacity

5,
2 at +94.0 in, 1 at +108.5 in, 1 at +127.0 in,
1 at +130.0 in

18. Emergency Exit

not recorded

19. Maximum Baggage/ Cargo Loads

227 kg (500 lb)
Two baggage compartments with max. floor loading of 21.1 kg/cm² (300 lb/ft²) and a total allowable load of 113.5 kg (250 lb) in each compartment.

20. Rotor Blade Control Movement
For rigging information refer to Maintenance Manual.

21. Auxiliary Power Unit (APU)
SOLAR T-62T-16A1 or T-62T-16A2

22. Life-limited Parts
refer to ALS

23. Wheels and Tyres
Tyres: not recorded
Wheels: not recorded

IV. Operating and Service Instructions

1. Flight Manual

Publication No. EAC007
The Airworthiness Limitations are listed in the S-64E Service Bulletins:
64B GENERAL-1 and 64B GENERAL-1 EASA Addendum, and 64B GENERAL-2

3. Service Letters and Service Bulletins
As published by Sikorsky Aircraft, Erickson Air-Crane Co., Erickson Air-Crane Co., L.L.C., Erickson Air-Crane Inc., DBA Erickson Air-Crane and Erickson Inc., DBA Erickson Air Crane.

Refer to approved RFM

5. Illustrated Parts Catalogue
n/a

6. Miscellaneous Manuals
not recorded

7. Required Equipment
The basic required equipment as prescribed in the applicable airworthiness regulations (see certification basis) must be installed in the aircraft for certification.
In addition, the following item of equipment is required:

V. Notes

1. Manufacturer’s eligible serial numbers:
64002, 64003, 64015 until 64019, 64022, 64025, 64027, 64028, 64033, 64034D, 64037, 64038, 64042, 64043, 64050, 64052, 64058, 64061, 64064 until 64066, 64079, 64099, 64101, 641001, 64E3001, and subsequent.

2. Type Certificate holder record note as per FAA TCDS H6EA, Revision 15, dated 17 March 2015:
- ‘Erickson Air-Crane Incorporated, DBA Erickson Air-Crane’ transferred TC H6EA to ‘Erickson Incorporated, DBA Erickson Air-Crane’ on 13 August 2014;
- ‘Erickson Air-Crane Co., L.L.C.’ transferred TC H6EA to ‘Erickson Air-Crane Incorporated, DBA Erickson Air-Crane’ on 14 February 2001;
- ‘Erickson Air-Crane Co.’ transferred TC H6EA to ‘Erickson Air-Crane Co., L.L.C.’ on 22 August 1997;

* * *
SECTION 3: OPERATIONAL SUITABILITY DATA (OSD)

The OSD elements listed below are approved by the European Aviation Safety Agency as per Commission Regulation (EU) 748/2012, as amended by Commission Regulation (EU) No 69/2014.

I. OSD Certification Basis

I.1 Reference Date for determining the applicable OSD requirements
reserved

I.2 MMEL - Certification Basis
reserved

I.3 Flight Crew Data - Certification Basis
reserved

I.4 SIM Data - Certification Basis
reserved

I.5 Maintenance Certifying Staff Data - Certification Basis
reserved

II. OSD Elements

II.1 MMEL
reserved

II.2 Flight Crew Data
reserved

II.3 SIM Data
reserved

II.4 Maintenance Certifying Staff Data
reserved
SECTION: ADMINISTRATIVE

I. Acronyms and Abbreviations

AEO  All Engines Operative          OEI  One Engine Inoperative
ALS  Airworthiness Limitations Section OSD  Operational Suitability Data
Amdt. Amendment                       P/N  Part Number
APU  Auxiliary Power Unit             PWR  Power
C.G.  Centre of Gravity               RFM  Rotorcraft Flight Manual
DA  Density Altitude                  rpm  Rounds Per Minute
EAC  Erickson Air-Crane               SC   Special Condition
ENAC  Ente Nazionale per l'Aviazione Civile shp  Shaft Horse Power
EU  European Union                    s/n  Serial Number
FAA  Federal Aviation Administration STA  Station
JAR  Joint Aviation Requirements     VFR  Visual Flight Rules
KIAS  Knots Indicated Air Speed       V_NE  Never Exceed Speed

II. Type Certificate Holder Record

II.1 Type Certificate Holder

<table>
<thead>
<tr>
<th>Type Certificate Holder</th>
<th>Period</th>
</tr>
</thead>
<tbody>
<tr>
<td>Sikorsky Aircraft Corporation 6900 Main Street</td>
<td>21 August 1969 until 13 February 1992</td>
</tr>
<tr>
<td>Stratford, Connecticut 06497, U.S.A.</td>
<td></td>
</tr>
<tr>
<td>Erickson Air-Crane Company 3100 Willow Springs Road,</td>
<td>14 February 1992 until 22 August 1997</td>
</tr>
<tr>
<td>Central Point, Oregon 97502, U.S.A.</td>
<td></td>
</tr>
<tr>
<td>Erickson Air-Crane Co., L.L.C. 3100 Willow Springs Road,</td>
<td>23 August 1997 until 14 February 2001</td>
</tr>
<tr>
<td>Central Point, Oregon 97502-0010, U.S.A.</td>
<td></td>
</tr>
<tr>
<td>Erickson Air-Crane Incorporated, DBA Erickson Air-Crane 3100 Willow Springs Road, Central Point, Oregon 97502-0010, U.S.A.</td>
<td>15 February 2001 until 12 August 2014</td>
</tr>
<tr>
<td>Erickson Incorporated, DBA Erickson Air-Crane 3100 Willow Springs Road, Central Point, Oregon 97502-0010, USA</td>
<td>Since 13 August 2014</td>
</tr>
</tbody>
</table>

III. Change Record

<table>
<thead>
<tr>
<th>Issue</th>
<th>Date</th>
<th>Changes</th>
<th>TC issue</th>
</tr>
</thead>
<tbody>
<tr>
<td>Issue 1</td>
<td>4 Aug 2004</td>
<td>EASA Initial Issue</td>
<td>Initial Issue, 4 August 2004</td>
</tr>
<tr>
<td>Issue 2</td>
<td>11 Nov 2005</td>
<td>- -</td>
<td>- -</td>
</tr>
<tr>
<td>Issue 3</td>
<td>27 Sep 2007</td>
<td>- -</td>
<td>Reissued, 27 September 2007</td>
</tr>
<tr>
<td>Issue 4</td>
<td>5 Feb 2013</td>
<td>- -</td>
<td>- -</td>
</tr>
<tr>
<td>Issue 5</td>
<td>23 Mar 2016</td>
<td>TC holder name updated; OSD data added; EASA TCDS format updated</td>
<td>Reissued, 23 March 2016</td>
</tr>
<tr>
<td>Issue 6</td>
<td>16 Dec 2021</td>
<td>SECTION 1: Emission reference, APU data, SBs and eligible s/n clarified SECTION 2: S-64E model added; EASA TCDS format updated</td>
<td>Reissued, 16 December 2021</td>
</tr>
</tbody>
</table>

- end of file -