Type Breezer B600



# **TYPE-CERTIFICATE**

# **DATA SHEET**

NO. EASA.A.598

for Breezer B600

Type Certificate Holder Breezer Aircraft GmbH & Co. KG

> Sönke-Nissen-Koog 58 25821 Reußenköge Germany

For models: B600

\*\*\*\* \* \* \*\*\* TE.CERT.00135-001 © European Union Aviation Safety Agency, 2021. All rights reserved. ISO9001 Certified. Page 1 of 8 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

Intentionally left blank



TE.CERT.00135-001 © European Union Aviation Safety Agency, 2021. All rights reserved. ISO9001 Certified. Page 2 of 8 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

# Contents

Section	A: Breezer B600	4
A.I	General	4
A.II	EASA Certification Basis	4
A.III	Technical Characteristics and Operational Limitations	5
A.IV	Operating and Service Instructions	7
A.V	Notes	7
Section	B: Administrative Section	8
B.I	Acronyms & Abbreviations	8
B.II	Type Certificate Holder Record	8
B.III	Change Record	8



TE.CERT.00135-001 © European Union Aviation Safety Agency, 2021. All rights reserved. ISO9001 Certified. Page 3 of 8 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

### Section A: Breezer B600

#### A.I General

- 1. Type/ Model/ Variant
  - 1.1 Type: Breezer B600
  - 1.2 Model: B600
  - 1.3 Variant: N/A
- 2. Airworthiness Category: Restricted
- 3. Manufacturer: Breezer Aircraft GmbH & Co. KG
- 4. EASA Type Certification Application Date: 10 September 2012
- 5. EASA Type Certification Date: 22 January 2016

### A.II EASA Certification Basis

1.	Reference Date for determining the applicable requirements:		10 September 2012
2.	Airworthiness Requirements:	Certification Specifications for Light Spor Initial Issue, June 27, 2011 For detailed Information see CRI-A01	t Aeroplanes CS-LSA,
3.	Special Conditions:	none	
4.	Exemptions:	none	
5.	Elected to Comply:	CS-23 acc. CC: 001-M-12-E-CC	
6.	Equivalent Safety Findings:	none	
7.	Environmental Protection:	ICAO Annex 16, Volume I (for more detai A.598)	ls see EASA TCDSN

#### A.III Technical Characteristics and Operational Limitations

1.	Type Design Definition:	001-M-12-MZ-TDD, Issue 1, Revision 2 or later approved revisions.			
2.	Description:	Single engine, two seats side-by-side, all metal, low-wing, fixed tricycle landing gear with steerable nose wheel.			
3.	Equipment:	For Minimum Equipment List refer to POH, Chapter II.11			
4.	Dimensions:	Wingspan	8.74 m	(344 in)	
		Length	6.73 m	(265 in)	
		Height	2.14 m	(84.3 in)	
		Wing area	11.89 m²	(127.97 sq ft)	

#### 5. Engine

- 5.1 Model: ROTAX 912 ULS2
- 5.2 Type Certificate: Certified as part of the aircraft
- 5.3 Limitations:

		Power:	Max. take-off: 73,5 kW at 5800 rpm (max. 5 min)
		Oil pressure:	Min: 0,8 bar
			Normal: 2,0 – 5,0 bar
			Max: 7,0 bar cold engine starting
		Oil temperature:	Max. 130 °C
			Min. 50 °C
		Fuel pressure:	Max. 0,4 bar
			Min. 0,15 bar
		Coolant temp.	Max. 120 °C
5.4	Maximum Continuous	Power:	69,0 kW at 5500 rpm
ved)			
, 			

- (Reserved)
  Propeller
- Propeller 1
  - 7.1 Model: NEUFORM CR3-V-70-(IP)-R2-ECS
  - 7.2 Type Certificate: Certified as part of the aircraft
    - 7.3 Number of blades 3 blades
    - 7.4 Diameter: 1750 mm
    - 7.5 Sense of Rotation: Clockwise in flight direction

# Propeller 2

- 7.6 Model: NEUFORM CR3-75-(IP)-47-101,6
- 7.7 Type Certificate: Certified as part of the aircraft
- 7.8 Number of blades: 3 blades
- 7.9 Diameter: 1750 mm
- 7.10 Sense of Rotation: Clockwise in flight direction



8.	Fluids							
	8.1	Fuel:			) LL. Refer	to ROT	MOGAS, AVGAS UL 91 or AX Operating Manual. rohibited.	
	8.2	Oil:		min. grade	API "SG" nditions. F	or high	erating Manual and SI-912-016 er, chose viscosity according to nended oil brands listed in	
	8.3	Coolant:		water as p	er ROTAX	Operat	vith a mixture ratio of 1:1 with ing Manual and SI-912-016. Is listed in ROTAX SI-912-016.	
9.	Fluid capaci	ties						
	9.1	Fuel:						
		Max. capacit	у	76 litres				
		Max. usable		74 litres				
	9.2	Oil:						
		Min.		approx. 2.5				
		Max		approx. 3.0				
		Coolant syste	•			2S		
10.	Air Speeds:		126 km/	-	3 KIAS)			
			179 km/	•	7 KIAS)			
		V <sub>NO</sub> :	197 km/	h IAS (10	06 KIAS)			
		V <sub>NE</sub> :	252 km/	h IAS (13	36 KIAS)			
11.	Maximum C	Certified Opera	ating Alti	tude:	13.	000 ft		
12.	Approved O	perations Cap	ability:		VF	R Day		
13.	Maximum N	Aasses						
	13.1	. Maximum Ta	ke-off M	ass: 600 kg	g (1320 lbs	5)		
	13.2	2 Maximum La	nding Ma	ass: 600 kg	g (1320 lbs	5)		
14.	Centre of G							
		Empty weigh						
		Flight:		Rear limit: Forward lim	300 it: 269		(22.0 % MAC) (19.8 % MAC)	
		i ligitt.		Rear limit:	435		(32.0 % MAC)	
15.	Datum:						leading edge.	
16.	(Reserved)					U	5 5	
	Levelling M	eans:	A	A Horizontal	plain thro	ough th	e upper edge of the canopy	
	0			guardrail	•	0		
18.	Minimum F	light Crew:	1	L (Pilot)				
19.	Maximum P	assenger Seat	ing Capa	city: 1				
20.	Baggage/ Ca	argo Comparti		The baggage Max. baggag	-		located behind the seats. (33 lbs)	
21.	Wheel and <sup>-</sup>	Tyres:		Main wheel Nose wheel		5.00- 4.00-		

22. (Reserved)

\*\*\*\* \* \* \*\*\*\* TE.CERT.00135-001 © European Union Aviation Safety Agency, 2021. All rights reserved. ISO9001 Certified. Page 6 of 8 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

#### A.IV Operating and Service Instructions

1.	Flight Manual:	English: POH-B600-A-E German: POH-B600-A-I From S/N 042LSA or mo POH-B600-E, POH-B600	D(*), POH-B600-E-D(*) odified according to 001	-Ä-19:
2.	Maintenance Manual:	MM-B600-E(*)		
3.	Structural Repair Manual:	Refer to Maintenance I	Manual	
4.	Weight and Balance Manual:	Refer to Flight Manual		
5.	Illustrated parts Catalogue:	N/A		
6.	Operating Manual and Maintena	ance Manual for Engine:	Operating Manual:	OM-912
			Maintenance Manual:	MMH-912,
				MML-912
7.	Operating Manual and Maintena	ance Manual for Propelle	r: NEUFORM-Adjustable Operating manual, NEUFORM–3-Blade-V Propeller R2 Series O	ariable Pitch

(\*) POH: or all later approved versions; (\*) MM: or all later accepted versions

### A.V <u>Notes</u>

N/A



TE.CERT.00135-001 © European Union Aviation Safety Agency, 2021. All rights reserved. ISO9001 Certified. Page 7 of 8 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

# Section B: <u>Administrative Section</u>

# B.I Acronyms & Abbreviations

CS-LSA	Certification Specification for Light Sport Aeroplanes
EASA	European Aviation Safety Agency
IAS	Indicated Airspeed
KIAS	Indicated Airspeed [knots]
LSA	Light Sport Aeroplane
MAC	Mean Aerodynamic Chord
TCDS	Type Certificate Data Sheet

### B.II Type Certificate Holder Record

Breezer Aircraft GmbH & Co. KG Sönke-Nissen-Koog 58 25821 Reußenköge Germany

#### B.III Change Record

Issue	Date	Changes	TC Issue No. & Date
01	22 January 2016	Initial Issue	Initial Issue,
			22 January 2016
02	30 November 2021	Introduction of one POH for all LSA with previous sales	-
		designation Elegance and Attraction, merging of all	
		product attributes	

-END-



TE.CERT.00135-001 © European Union Aviation Safety Agency, 2021. All rights reserved. ISO9001 Certified. Page 8 of 8 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.