TYPE-CERTIFICATE DATA SHEET

No. P.177

For Propeller
KW-3(x) series propellers

Type Certificate Holder
Aleš KŘEMEN
Vodolská 4, Dolínek
250 70 Odolena Voda
Czech Republic

For Models:
KW-30
KW-31
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I. General

1. Type/ Models
   KW-3(x) / KW-30, KW-31

2. Manufacturer
   Aleš KŘEMEN
   Vodolská 4, Dolinek
   250 70 Odolena Voda
   Czech Republic

3. Date of Application

<table>
<thead>
<tr>
<th>Type</th>
<th>Date</th>
</tr>
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<tbody>
<tr>
<td>KW-30</td>
<td>01 March 2013</td>
</tr>
<tr>
<td>KW-31</td>
<td>01 March 2013</td>
</tr>
</tbody>
</table>

4. EASA Type Certification Date

<table>
<thead>
<tr>
<th>Type</th>
<th>Date</th>
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<tbody>
<tr>
<td>KW-30</td>
<td>18 February 2014</td>
</tr>
<tr>
<td>KW-31</td>
<td>18 February 2014</td>
</tr>
</tbody>
</table>

II. Certification Basis

1. State of Design Authority Certification Basis
   Czech Republic

2. Reference Date for determining the applicable airworthiness requirements
   01 March 2013

3. EASA Certification Basis

   3.1. Airworthiness Standards

   3.2. Special Conditions
      None

   3.3. Equivalent Safety Findings
      None

   3.4. Deviations
      None
III. Technical Characteristics

1. Type Design Definition
Each design configuration is defined by a main assembly drawing and an appropriate parts list.

The KW-30 propeller model covers the following design configurations.
Design Configuration “Constant Speed”
Drawing No. 30-000-000 dated 21 January 2014 (*1)
Above mentioned drawing contains Parts List

The KW-31 propeller model covers the following design configurations.
Design Configuration “Constant Speed”
Drawing No. 31-000-000 dated 21 January 2014 (*1)
Above mentioned drawing contains Parts List

(*1) effective is the declared issue or a later approved revision.

2. Description
The KW-30, resp. KW-31, propeller models are a 3-blade hydraulically, resp. electrically controlled variable pitch propellers. Provided that the power unit is equipped by hydraulic, resp. electric governor the propellers could work as a constant speed propeller.
The hub is milled out of aluminium alloy and the blades are made of wood with glass or carbon composite covering layer. The leading edge of the blade is protected by the metal stamping.

3. Equipment
Spinner: according to Aleš KŘEMEN Service Bulletin No. 2
Governor: according to Aleš KŘEMEN Service Bulletin No. 3

4. Dimensions
Propeller diameter: max. 195 cm (77”)

5. Weight
KW-30: approx. 12.5 kg (27.6 lb)
KW-31: approx. 14.0 kg (30.9 lb)

6. Hub/Blade-Combinations

<table>
<thead>
<tr>
<th>Hub</th>
<th>Blade - Type</th>
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<tbody>
<tr>
<td>KW-30</td>
<td>-031, -033, -034, -037, -038</td>
</tr>
<tr>
<td>KW-31</td>
<td>-031, -033, -034, -037, -038</td>
</tr>
</tbody>
</table>

7. Control System
Propeller governors as listed in Aleš KŘEMEN Service Bulletin No. 3.

8. Adaptation to Engine
Adaptation to engine as listed in Aleš KŘEMEN Service Bulletin No. 4.
9. Direction of Rotation
Direction of rotation (viewed in flight direction) as identified by a letter-code in the propeller designation. (see chapter VI.3.). The left-hand version of an approved propeller model is approved at the same rating and diameter as listed for the right-hand model.

IV. Operating Limitations

1. Maximum Take Off Power and Speed

104 kW
- all blades except Type -037, -038: max. 2552 min-1
- blades Type -037, -038: max. 2283 min-1

117 kW
- blades Type -037: max. 2283 min-1
- blades Type -038: max. 2323 min-1 - max. 1min

2. Maximum Continuous Power and Speed

104 kW
- all blades except Type -037, -038: max. 2552 min-1
- blades Type -037, -038: max. 2283 min-1

117 kW
- blades Type -037, -038: max. 2283 min-1

3. Propeller Pitch Angle
Maximum pitch change range +5° to +50° - measured at 75% radius station

V. Operating and Service Instructions

<table>
<thead>
<tr>
<th>User’s Manual</th>
<th>KW-30</th>
<th>UM-06, Rev.0, dated 21 January 2014 [*]</th>
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<tr>
<td></td>
<td>KW-31</td>
<td>UM-05, Rev.0, dated 21 January 2014 [*]</td>
</tr>
<tr>
<td>Overhaul Manual</td>
<td>KW-30</td>
<td>TN-21, Rev. 0, dated 21 January 2014 [*]</td>
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<tr>
<td></td>
<td>KW-31</td>
<td>TN-22, Rev. 0, dated 21 January 2014 [*]</td>
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<tr>
<td>Service Bulletins</td>
<td>KW-3(x)</td>
<td>as noted in the current List of Service Bulletins</td>
</tr>
</tbody>
</table>

[*] or later approved revision
VI. Notes

1. The EASA approved Airworthiness Limitations Section (ALS) of the Instructions for Continued Airworthiness is published in the applicable “User Manual” document, chapter 17 “Airworthiness Limitations Section”. This ALS section is empty because no life limit is necessary for these models.

2. The overhaul intervals recommended by the manufacturer are listed in Aleš KŘEMEN Service Bulletin No. 1.

3. **Propeller designation system:**

   **HUB:**

   1. **KW**  Aleš KŘEMEN, Vodolská 4, Dolínek, 250 70 Odolena Voda, Czech Republic
   2. No. of propeller model
   3. Code letter for propeller category:
      - A - Automatic Propeller
      - F - Fixed Pitch Propeller
      - G - Ground Adjustable Propeller
      - V - Variable Pitch Propeller
   4. Code letter for blade pitch change system:
      - H - Hydraulic
      - E - Electric
      - M - Mechanical
   5. Number of blades installed
   6. Code letter for feathering system:
      - F – Feather position installed
      - 0 – No feather position possible
   7. Code letter for reverse provision:
      - R – Reverse position installed
      - 0 – No reverse position possible
   8. Code letter for flange type
      - listed in Aleš KŘEMEN Service Bulletin No. 4

   **BLADE:**

   9. Code letter for blade design and installation:
      - R:  - Right-hand tractor
      - RP: - Right-hand pusher
      - L:  - Left-hand tractor
      - LP: - Left-hand pusher
   10. Propeller diameter in cm
   11. No. of blade type (contains design configuration and aerodynamic data) according to the certified hub/blade-combinations.
SECTION: ADMINISTRATIVE

I. Acronyms and Abbreviations
n/a

II. Type Certificate Holder Record
n/a

III. Change Record

<table>
<thead>
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<th>Issue</th>
<th>Date</th>
<th>Changes</th>
<th>TC issue</th>
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<tbody>
<tr>
<td>Issue 01</td>
<td>18 February 2014</td>
<td>Initial Issue</td>
<td>Initial Issue, 18 February 2014</td>
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<tr>
<td>Issue 02</td>
<td>26 June 2018</td>
<td>Increase of maximum take-off power and max. continuous power (Major Change Approval 10065886)</td>
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<tr>
<td>Issued 03</td>
<td>06 May 2022</td>
<td>Increase of maximum take-off power and max. continuous power (Major Change Approval 10079112). Certification Basis changed from CS-P Amendment 1 to CS-P Amendment 2.</td>
<td>TC certificate reissued to correct the address of the applicant. 06 May 2022</td>
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