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<td>10013930</td>
<td>AERODATA AG</td>
<td>REV. 2</td>
<td>B300</td>
<td>Installation of Aerodata Automatic Flight Inspection System AD-AFIS-112.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. CS-23, Amendment 2. (selected to comply by the applicant, not required by 21A.101). The requirements for environmental protection and the associated Technical Documentation.</td>
<td>Installation Manual: ..., HB-ACC-1087, Rev. (Master Document List: ..., MLO/4/ C-067, Rev. B). Maintenance Manual Supplement, ..., AMM-1088, Rev. 8. Flight Manual Supplement: ..., AFM-124, Rev. 1, or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>06/05/2011</td>
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<td>10013945</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 10</td>
<td>LBA TA0441 ISSUE 4</td>
<td>Conversion of A300B4-600 and A300B4-600SR aircraft from a Passengerversion to a Freighter version (PDF conversion).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any amendments.</td>
<td>-. Certification Summary Document EAA2400000/001/COS Issue 00, dated 16 January 2007. Airbus Deutschland Engineering Order No. PFM 09400A Issue 02, dated 07 May 2007. - Aircraft Flight Manual Supplement for Airbus A300-600, Revision 07 dated 15 January 2007. - Supplemental Airworthiness Limitations. In cases where an AMOC in the AMOC SUMMARY LIST is described as partial compliance to the AD requirements, the remaining mandatory actions must be performed as indicated within the airworthiness directive. Limitations mentioned in the A300-600P to F – Weight and Balance Manual Supplement.</td>
<td>None</td>
<td>06/07/2021</td>
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<td>10013954</td>
<td>TROYES AVIATION S.A.R.L.</td>
<td>REV. 3</td>
<td>CESSNA F177RG</td>
<td>Installation of a camera sight door under the fuselage.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. File ref: FAM N° 5000000000 CHANGE FILE N° 53 ISSUE JUNE 2006 - EASA.A.S.00494 R1 and ROMINOR CHANGE N° 41.</td>
<td>Applicability F177RG SN 0112, 0485, 103, 087 F-BURU and SN 51 F-HBAG.</td>
<td>29/03/2010</td>
<td>Active</td>
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<td>10013960</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 6</td>
<td>A310-304/-308/-324/-325 A310-203/-221/-222</td>
<td>EASA STC 10013960 Revision 6 (A310-200 series and -300 series aeroplanes, Passenger to Freighter conversion) (Ref. to former Luftfahrt Bundesamt STC TA0383 issue 9): The STC EASA STC 00064 has covered the transfer of the STC TA0383 issue 9 (original Aviation Authority reference) from the original AIRBUS.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. See Certification Summary Document referenced in Associated Technical Documentation section.</td>
<td>File ref: FAM N° 5000000000 CHANGE FILE N° 53 ISSUE JUNE 2006 - EASA.A.S.00494 R1 and ROMINOR CHANGE N° 41.</td>
<td>None</td>
<td>27/06/2012</td>
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<td>10013963</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 3</td>
<td>767-300</td>
<td>Introduction of Matuwha In-Flight Entertainment System Rev.3: Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer. Prior to installation of this change/repair it must be determined that...</td>
<td>Aerospace Engineering Solutions Master Data List AES-757-357MDL or later approved revision.</td>
<td>28/04/2021</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<th>EASA STC No.</th>
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<tr>
<td>10013964</td>
<td>LUTHANSA TECHNIK AG</td>
<td>REV. 3</td>
<td>A340</td>
<td>Installation of non-electrical Floor Path Marking System</td>
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<td></td>
<td>STC TAS-0105 Supplemental Type Design Document STD203-9701M, issue 07 dated 09.11.2006. Applicant's Declaration of Compliance dated 15.11.2006. Airplane Flight Manual Supplement A-33-06/27-4FM. Instructions for Continued Airworthiness. later revisions of the above listed documents approved by...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/02/2014</td>
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<td>10013967</td>
<td>LUTHANSA TECHNIK AG</td>
<td>REV. 3</td>
<td>747-300, 747-400, 747-8</td>
<td>Radome installation for Live TV antennaRev. 1: Incorporation of B747-300 in the applicable listRev. 2: Incorporation of B747-8 in the appilability listSTC Original number EASA.A.S.00701 replaced by new number 10013967Rev. 3: Extension of 747-8 ALI taek, based on reassessment of DTAnalysis w...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>B747-300 Supplemental Type Design Document STD203-1220M Issue 01 rev 05Maintenance requirements document C-AAL006-MR rev EASA.740-400 Supplemental Type Design Document STD203-1220M Issue 01 rev 05Instruction for Continued Airworthiness ICA53-0215-01 rev EASA-740-4 Supplemental Type Design Document STC...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2020</td>
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<td>10013970</td>
<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td>REV. 3</td>
<td>B200</td>
<td>Photo Conversion (creation of two hours for camera) Rev. 3: Transfer of EASA.A.S.00741 Issue 1 for S/N BB-1181 and BB-1607 Revision 2 for S/N BB-221 and BB-1607</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>For S/N BB-1181: EASA-16-PNF-045 Rev 3 dated 18/12/2008 For S/N BB-221 and BB-1607: EASA-16-MLD-F050 Rev 1 dated 17/12/2008 EASA-16-AMT-F050 Rev 3 dated 18/12/2008 EASA-16-AMM-F050 Rev 1 dated 17/12/2008 EISA-16-ICA-F050 Rev 1 dated 17/12/2008 later revisions of the above lists...</td>
<td>Applicable to S/N B200 S/N BB-1181, BB-221, BB-1607 Identification of intercom prohibited in dense traffic areas Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/04/2024</td>
<td>Active</td>
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<td>10013975</td>
<td>SMA AERO ENGINES GMBH</td>
<td>REV. 15</td>
<td>F162P, F182QB</td>
<td>Installation of SMA Diesel Engine SR360-230 or SR360-230G and MT-Propeller MTV-051B-1809-08 Rev. 14: Transfer of STC 10013975 rev 12 from Spanish Aircraft Engines S.A. to SMA Aero Engines GmbHRev. 15: Update of annex 1 for tips and incorrect references.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any in...</td>
<td>- SMA Mod. AM 16-082- Flight Manual Supplemets: see Appendix dated 17/02/2021 - Airworthiness Limitation Information asper Aircraft Service ManualC182-ASMS-01E Rev. 6. Section 23MAL - Installation Instructions: see Appendix dated 17/02/2021 - Aircraft Service ManualC182-ASMS-01E latest revision...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/06/2023</td>
<td>Active</td>
<td>1013975</td>
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<td>10013976</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 2</td>
<td>5550 (CITATION SI) S560 (CITATION ULTRA) SS50 (CITATION II)</td>
<td>Installation of an L3 Digital Flight Data RecorderRevision 2 was raised due to missing technical document references in Revision 1</td>
<td>EASA TCDD.IMA.207Hi The environmental protection and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: JAR 25 at change 15The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Master Documents List 2018778-01.DOC Issue 8 dated 27 Sept. 2010 - Engineering Order 2018778-EO-01 Issue 5 dated 26 Sept. 2010 - Aircraft Flight Manual Supplement 2018778 AM-01A, Issue 5 from 16/January 2010 for 115 VAC or alternative Aircraft Flight Manual Supplement 2018778-AM-01B, Issue 2 from 19...</td>
<td>Limited to S/N 550-0001 through 550-0733 (Citation II) S/N 550-0001 through 550-1136 (Citation Bravo) S/N 560-0001 through 560-0259 (Citation VI) S/N 560-0260 through 560-0538 (Citation Ultra) S/N 6550-0001 through 6550-0160 (Citation SI II)</td>
<td>05/10/2010</td>
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<td>Limitations/Conditions</td>
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<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
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<td>737-700</td>
<td>Installation of Enhanced Mode S TranspondersRev. 1: Transfer of STC from SR Technics Switzerland to groWING Engineering Partners GmbH</td>
<td>- EASA STC no. 737-700 issued 2 November 2005 or later approved revision Flight Manual Supplement BBJ/M/0514 Issue 6</td>
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<td>10013993</td>
<td>AERODATA AG</td>
<td>REV.11</td>
<td></td>
<td>DORNIER 328-100</td>
<td>Installation of Maritime Surveillance System on Dornier 328-100 using SAR, FLIR, and optional IR/UV Scanner Revision 4 Functional optional upgrade to the radar system and mission software to optimize search and rescue operation. Hardware upgrade to components of ELTA weather and mission radar.</td>
<td>- Master Document List ref MDL/AV-089 Rev. 13. Flight Manual Supplement ref AFM-077 Rev. 10. - AERODATA Maintenance Manual Supplement AMC-782 Rev. 6 dated September 2012 or later revisions of the above listed documents approved by EASA</td>
<td>Applicable to Aircraft Serials 3051, 3052, 3053, 3057 and 3059 as modified by Aerodata Modification D328.2.xx for AMSA Mission Purpose. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/01/2013</td>
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<td>10014001</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>757-200</td>
<td>Installation of Enhanced Mode S TranspondersRev. 1: Transfer of STC from SR Technics Switzerland to groWING Engineering Partners GmbH</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Aviation Traders Ltd Definition Document DD/0219/02 Issue 2-Aviation Traders Ltd AFM Supplement Number 31 Issue 1 or later revisions of the above listed document (s) approved/accepts the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/11/2021</td>
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<td>10014002</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>737-700</td>
<td>Installation of Enhanced Mode S TranspondersRev. 1: Transfer of STC from SR Technics Switzerland to groWING Engineering Partners GmbH</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Aviation Traders Ltd Definition Document DD/0229/18 Issued Aviation Traders Ltd AFM Supplement Number 29 Issue 1 or later revisions of the above listed document (s) approved/accepts the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/11/2021</td>
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<td>10014003</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>737-300</td>
<td>Installation of Enhanced Mode S TranspondersRev. 1: Transfer of STC from SR Technics Switzerland to groWING Engineering Partners GmbH</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Aviation Traders Ltd Definition Document DD/0239/29/ Issue 2-Aviation Traders Ltd AFM Supplement Number 30 Issue 1 or later revisions of the above listed document (s) approved/accepts the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10014007</td>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 2</td>
<td></td>
<td>B747-400</td>
<td>Reconfiguration of Virgin Atlantic Airways B747-400 from LHR to LDN configuration in accordance with SR Technics Master Data List CT747-25-1001-MDL-01, Rev 3 or later approved revisionRev. 2: Transfer of STC from SR Technics Switzerland to groWING Engineering Partners GmbH</td>
<td>The Certification Basis for the original product remains unchanged and applicable to this certificate/approval.</td>
<td>- CT747-25-1001-CCS, Rev 3 CT747-25-1001-MDL-01, Rev 3 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/04/2021</td>
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<td>Airbus Engineering GmbH</td>
<td>PA-23-250</td>
<td>Installation of COM/NAV/ILS/GPS- System Garmin GNS 4302, ELT Kannad 406AF</td>
<td>Engineering Order 003-04 Rev 1EASA approved FMS Emergency Locator Transmitter Kannad 406AF Doc No PD011-000406 Rev 0 or later EASA approved issue</td>
<td></td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>17/02/2021</td>
<td>Active</td>
<td>10014011</td>
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<td>Babcock Mission Critical Services FM Management S.A.U.</td>
<td>A330-111/-112/-113/-114/-115</td>
<td>Change to the complete avionics set installed by a new one last generation GARMIN and HONEYWELL</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td><em>Documento Técnico CMA014, issue 1</em></td>
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<td>15/02/2019</td>
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<td>Luftansa Technik AG</td>
<td>A320-271N/-272N/-273N</td>
<td>Installation of non-electrical Floor Path Marking System - Configuration 1: All models (except NEO series). Configuration 2: Guidetaxi 1000, all models; Revision 01 and 02: Editorial corrections. Revisions 03: Extended applicability (A320 family NEO models)</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following paragraph (s) at a later amendment.</td>
<td>Supplemental Type Design Definition STDO3-04/03M issue 03 rev. 0</td>
<td>Restricted to: s/n FTB3370020, FTB3370022, FTB3370023, FTB3370025</td>
<td>04/04/2023</td>
<td>Active</td>
<td>10014013</td>
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<td>Airbus S.A.S.</td>
<td>A319-133</td>
<td>Airbus Corporate Jet Centre Modification CJ0072 - AL SALAM - CONVERSION/PRIVATE TO PUBLIC AND VICE VERSA.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements as listed in the General CS 25.307(a)(d), 25.301(a)</td>
<td>Modification Approval sheet CJ0072 Issue 01 - CONVERSION PRIVATE TOPUBLIC AND VICE VERSA, issue 01 dated 38th May 2016. AFM Supplement P&amp;O MOD SOG-06-1030 dated MAY 01/1665 later revisions of the above listed documents approved by EASA.</td>
<td>This STC approval is limited to A318-133 MSN 2421 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/05/2016</td>
<td>Active</td>
<td>10014023</td>
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<td>Ruag Aerospace Services GmbH</td>
<td>Cessna 550</td>
<td>Introduction of TCAS II Change 7.1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Document List, Ruag Aerospace document No. - 0147-MLD, Revision 0 dated 9 February 2015 Aircraft Flight Manual Supplement, Ruag Aerospace document No.-0147- AFMS. Revision 0 dated 6 February 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>20/02/2015</td>
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<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10014040</td>
<td>SAFRAN SEATS</td>
<td>REV. 1</td>
<td>767-300</td>
<td>Introduction of Stowage 4B and infill panel to British Airways aircraft as defined in the modification definition document and Declaration of Design and Performance (DDP)</td>
<td>- The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Health Tecna Modification document C035 and associated DDP dated 26/01/2005 or later approved revisions</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/03/2019</td>
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<td>10014046</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 1</td>
<td>CESSNA 208, 208B</td>
<td>1. Installation of the MTV-16-1-E-C-R (P)/CFR200-55 full feathering and removable 4-blade constant speed propeller2. Extension of installation 1 on MTV-16-1-E-C-R (P)/CFR200-55propellers equipped with fluid device system.</td>
<td>- The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval FAR 23 including amendment 34 (not required by 21A.101, elected to comply by the applicant). This certificate/ approval involves a change to...</td>
<td>Installation 1–Instructions for Continued Amorworthiness and Installation Instructions/ E-1166- Airplane Flight Manual Supplement No E-1166/Installation 2–Instructions for Continued Amorworthiness</td>
<td>None</td>
<td>12/04/2011</td>
<td>Active</td>
<td>10014046</td>
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<td>10014049</td>
<td>BUSINESS AND COMPUTER AIRCRAFT</td>
<td>REV. 1</td>
<td>BEECH B300</td>
<td>Installation of KLN 900 GPS with EFIS-85 according to STCEASA-A.S.0.01131.</td>
<td>- The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval FAR 23 Amendment 23.42/23.42. The certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>File ref: BCA 26 Rev 2 dated 03/10/2011 AFM SUPPLEMENT FOR GPS BENDEX / KING KLN-900 Ed. 01/12 - PIN BCA26 Rev25MV.</td>
<td>BEECH 300 B300 - F-GPRH - SN FA 226.</td>
<td>06/02/2012</td>
<td>Active</td>
<td>10014049</td>
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<td>10014059</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 3</td>
<td>757-200</td>
<td>Introduction of Pegasus Flight Management Computers &amp; MultiMode Receiving (MMR)/Rv 3. Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>- The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-25 Initial Amendment 25.589(a)(6): 25.601, 25.603, 25.605(a), 25.605(b), 25.606, 25.609(a), 25.613(a), 25.614(a), 25.617(a), 25.618(a), 25.619(a), 25.689(a)(4), 25.699.</td>
<td>AES Flight Manual Supplement No. 71 issue 2 and/or later revisions of the above listed documents approved by EASA/ AES Modification No. AES-757-378 issue AES Master Data List AES-757-379MD, Issue 15</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/04/2021</td>
<td>Active</td>
<td>10014059</td>
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<tr>
<td>10014066</td>
<td>MAGNETIC MRO AS</td>
<td>REV. 2</td>
<td>747-400</td>
<td>Modification of Cabin to Install Zone B Cabin Class Dividers, Zone Estructo-665 &amp; Dogbox, Door 2 Port &amp; Starboard Dividers &amp; Upper Deck Dogbox. Revision: Transfer of Certificate to Magnetic MRO AS.</td>
<td>- The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Modification Leaflet 04ML110.</td>
<td>1. Compliance with the following requirements has not been established: 1.1. Viewing of all applicable placards &amp; signage to ensure that FAR/JAR 25.791.1.2. Weight, Centre of Gravity &amp; Weight Distribution, to be in compliance with JAR25.1519.1.2. Compliance with JAR-FAR 25.791.1.2. The elimin...</td>
<td>19/09/2019</td>
<td>Active</td>
<td>10014066</td>
<td></td>
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<tr>
<td>10014074</td>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 2</td>
<td>JETSTREAM 3200</td>
<td>Installation of Airborne Collision Avoidance System (ACASII).</td>
<td>- Avionics Mobile Services Definition Document Part 21-002/1485-01 Issued or later approved revision.</td>
<td>- Avionics Mobile Services AFM Supplement Part 21-008/011 Issue 1 later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>27/06/2018</td>
<td>Active</td>
<td>10014074</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAAC STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10014078</td>
<td>SAFRAN SEATS</td>
<td>REV. 1</td>
<td>747-400</td>
<td></td>
<td>Reconfiguration of Cabin interior from current 387SA passenger cabin configuration (169-158P-113Y for RT 514) to a 383 (129+47B+34Y). Cathy Pacific Bosing 747-400 fleet standard passenger cabin configuration.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. Thererequirements for environmental protection and the associatedcertified noise and/or emissions levels of the original pro...</td>
<td>Health Tecra Modification Document C933 and associated DDP dated 21/6/05 or later approved revisions.</td>
<td>As defined in DDP. Aircraft Serial MSN 26548, Zero Occupancy. For this modification to be accepted on an EU registered aircraft, the operator must confirm that all the associated FAA and Hong Kong CDBmodifications have been EASA approved. Prior to installation of this design change it must be det...</td>
<td>14/02/2019</td>
<td>Active</td>
<td>10014175</td>
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<tr>
<td>10014090</td>
<td>ROEDER PRIVATGEBUSGMBH</td>
<td>REV. 1</td>
<td>SA277-BC, SA277-DC, SA277-TC, SA277-AC, SA277-AT, SA262-AT, SA262-T(B), SA262-T, Installation of Thrommen Air Data Display and Attenuator.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement Doc No SA262/227, NY-2012-06, Rev. 0, or later EASA approved revision.</td>
<td>Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/or repair will introduce a new adverse effect upon the airworthiness of the product.</td>
<td>25/09/2017</td>
<td>Active</td>
<td>10014080</td>
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<td>10014097</td>
<td>SAFRAN SEATS</td>
<td>REV. 2</td>
<td>767-300</td>
<td></td>
<td>Introduction G3 Wardrobe to British Airways aircraft as defined in the modification definition document and Declaration of Design and Performance (DDP)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. Thererequirements for environmental protection and the associatedcertified noise and/or emissions levels of the original pro...</td>
<td>Health Tecra Modification Document C936 and associated DDP, dated 24/March 2005, or later approved revisions.</td>
<td>As defined in the DDP. Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/or repair will introduce a new adverse effect upon the airworthiness of the product.</td>
<td>14/02/2019</td>
<td>Active</td>
<td>10014097</td>
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<tr>
<td>10014122</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 2</td>
<td>767-300</td>
<td></td>
<td>Introduction of 288 Passenger Configuration &amp; Associated Cabin Changes Rev. 1: FAR Part 26 Validation for EASA STC EASA.S.01232 Rev. 2. Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>AES Ltd. Master Data List No. AES.767-331MD, Issue 6 dated 14 December 2012, or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/or repair will introduce a new adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved o...</td>
<td>27/04/2021</td>
<td>Active</td>
<td>10014122</td>
</tr>
<tr>
<td>10014123</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 1</td>
<td>767-300</td>
<td></td>
<td>Introduction of Mitsubishi in-Flight Entertainment SystemRev 1. Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>AES Master Data List AES.767-351MD, Issue 2 or later EASA approved revision</td>
<td>Prior to installation of this change/repair it must be determined that there is an interrelationship between this change/repair and any other previously installed change/repair and/or repair will introduce a new adverse effect upon the airworthiness of the product.</td>
<td>28/04/2021</td>
<td>Active</td>
<td>10014123</td>
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<tr>
<td>10014125</td>
<td>SABENA TECHNICS BOC</td>
<td>REV. 1</td>
<td>A321-213/211/222, A321-151/211-212 A320-233 A320-211/212/214-231/232 A319-151/210-135 A319-111-210-112-114-115</td>
<td></td>
<td>ELT Antenne Installation/Installation of an Emergency Locator Transmitter external RAYAN antenna/PN: 2624-82, structuralelements only.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, ref: 0243-01-A-2101/ R2011Instructions for Continued Airworthiness, ref: 0243-01-A-2211/R20</td>
<td>Airworthiness Limitations inspection requirements apply for ExtendedService Goal aircraft, as described in section 3 of this modification toContinued Airworthiness (Supplement), ref: 0243-01-A-2211/R2. Prior to installation of this change/repair it must be determined that there is an interrelationship between th...</td>
<td>08/02/2019</td>
<td>Active</td>
<td>10014125</td>
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<tr>
<td>10014138</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 2</td>
<td>35-35A</td>
<td></td>
<td>Installation of Honeywell MST-87A dual Mode 8 VOR/DME/CAS-7A TCAS II System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>1. ACC Master Document List ACC-LR35-5006, Revision 0, or later EASA approved revisions 2. ACC AFMS LR35-5006-B20-S-MF, Revision 0, or later EASA approved revisions 3. ACC AFMS LR35-5006-B30-S-MF, Revision 0, or later EASA approved revisions. Prior to installation of this change/repair it must be determined that there is an interrelationship between this change/repair and any other previously installed change/repair and/or repair will introduce a new adverse effect upon the airworthiness of the product.</td>
<td>19/01/2022</td>
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<td>10014138</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
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<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10014130</td>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 2</td>
<td></td>
<td>560 AND 650</td>
<td>Installation of an Airborne Collision Avoidance System (ACAS II).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>This STC is approved only for the product configuration as defined in the associated design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>21/02/2018</td>
<td>Active</td>
<td>10014130</td>
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<tr>
<td>10014142</td>
<td>ROEDER PRÄZISION GmbH</td>
<td>REV. 2</td>
<td>A330-322</td>
<td>Cabin reconfiguration from a 18 B/C + 331 Y/C layout to a 18 B/C +331 Y/C layout. This STC installs a new cabin arrangement.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. As per EASA TCDS A.004/Elec to Comply with CS25 (Original Issue) requirements: 25.811(1)(v) 25.812(1)(c) 25.813 (a)(b)(c) 25.853(a)(3)(f) 25.865(a)(b)(c). The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>15/06/2018</td>
<td>Active</td>
<td>10014142</td>
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<tr>
<td>10014143</td>
<td>SAFRAN SEATS</td>
<td>REV. 1</td>
<td>747-400</td>
<td>Heath Tecna Modification C943 - Cathay Interiors Modification.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments (if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>As defined in DDP Aircraft Serial Numbers 25172 &amp; 25128. Zero Occupancy: For this modification to be accepted on an EU registered aircraft, the operator must confirm that all the associated FAA &amp; Hong Kong/CAD modifications have been EASA approved. Prior to installation of this design change it may be necessary to obtain a revision of the foreign approval.</td>
<td>14/03/2019</td>
<td>Active</td>
<td>10014143</td>
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<td>10014147</td>
<td>ATELIER CHABORD</td>
<td>REV. 3</td>
<td>DR 400/2D, DR 400/2D0, DR 400/120, DR 400/120 D, DR 400/120 D, DR 400/2, DR 400/100, DR 300/108, DR 315, DR 400/100, DR 300/108, DR 315</td>
<td>Installation of exhaust system according to Chabord modification STC-DR420-2015-01. Rev. 3: Enhanced silent block. Alternative anti corrosion surface treatment. Enhanced material.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements is applicable to this certificate/ approval. The change is limited to the aircraft as defined by Part 21, Point 21.109. Prior to installation of this change it must be determined that: the interrelationship between this change/repair and any other previously installed change/repair and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>The approval holder shall fulfill all the obligations of Part 21, Point 21.109. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/02/2012</td>
<td>Active</td>
<td>10014147</td>
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<td>10014148</td>
<td>SCANDINAVIAN AVIONICS DESIGN A/S</td>
<td>REV. 1</td>
<td>2, SA227-TT, 2, SA220-T, SA227-AT, 2A226-T, 2A226-AT, 1A227-BC, 1A227-BC, 1A226-TC, 2A226-AC</td>
<td>Installation of Honeywell CAS 87A CASAC II System with dual Mode STransponder</td>
<td>Design:Non-significant change. The Certification Basis of the original product and the following additional / alternative airworthiness requirements are applicable to this certificate/ approval. The change is limited to the aircraft as defined by Part 21, Point 21.109. Prior to installation of this change/repair it must be determined that: the interrelationship between this change/repair and any other previously installed change/repair and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>EASA approved Engineering Order, Doc. No. 207045-EO, Issue2, dated 07-Oct-2015, or later approved issue.</td>
<td>This change is limited to the configuration of aircraft / equipment types listed under Chapter 4.4 in EASA approved Engineering Order, Doc. No. 207045-EO, Issue 2, dated 25-Oct-2015, or later approved issue.</td>
<td>08/11/2010</td>
<td>Active</td>
<td>10014148</td>
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<td>10014159</td>
<td>TELEDYNE UK LIMITED</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING 767-300</td>
<td>Installation of Teledyne Controls QOAR (Controll Oxtoal Quick Access Recorder) Rev. 1. Transfer of Approval Certificate to new Certificate Holder (from Teledyne Ltd to Teledyne UK Ltd). Rev. 2: Correction of the Applicant name and address.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>The approval holder shall fulfill all the obligations of Part 21, Point 21.109. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/11/2020</td>
<td>Active</td>
<td>10014159</td>
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<td>10014160</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 1</td>
<td>F29MK500</td>
<td>Installation of an ARTEX ELT CAI86-1,2 Rev. 1. Transfer of STC from Gromotiq Aircraft Services GmbH to ACC Columbia Jet Service GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>None.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2022</td>
<td>Active</td>
<td>10014160</td>
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<td>STC holder</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>Mfr.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitation/Condition</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>767-300</td>
<td>Installation of Tidyline Flight Data Acquisition Unit P/N 233300-811 per provide parameters to the FORRRev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product characteristics applicable to this certificate/approval. The requirements for environmental protection and the associated certifiable noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;, Compatibility with other aircraft configurations shall be determined by the installer. Prior to installation of this change/repair it must be determined that...</td>
<td>09/11/2021</td>
<td>Active</td>
<td>10014163</td>
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<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>114B</td>
<td>Wing auxiliary Spar RepairRev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product characteristics applicable to this certificate/approval. The requirements for environmental protection and the associated certifiable noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification Summary Sheet - ALT/COM/M001 or later revision of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
<td>Active</td>
<td>10014175</td>
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<tr>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 2</td>
<td></td>
<td>HAWKER 800XP</td>
<td>Installation of an Airborne Collision Avoidance System (ACAS II).</td>
<td>- Avionics Mobile Services Project Certification Plan AM1498 Reference/PART2-003/149 8-01 or later approved revision - Avionics Mobile Services AFM Supplement reference AFM PART2-008 Issue 2 or later approved revision.</td>
<td>- Avionics Mobile Services Project Certification Plan AM1498 Reference/PART2-003/149 8-01 or later approved revision - Avionics Mobile Services AFM Supplement reference AFM PART2-008 Issue 2 or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;, Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>27/08/2018</td>
<td>Active</td>
<td>10014184</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td></td>
<td>767 SERIES</td>
<td>Installation of non-electrical Floor Path Marking System.</td>
<td></td>
<td></td>
<td></td>
<td>05/03/2014</td>
<td>Active</td>
<td>10014191</td>
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<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 3</td>
<td></td>
<td>HAWKER 800XP</td>
<td>Installation of an Airborne Collision Avoidance System (ACAS II).</td>
<td>- Avionics Mobile Services Project Certification Plan AM1498 Reference/PART2-003/149 8-01 or later approved revision - Avionics Mobile Services AFM Supplement reference AFM PART2-008 Issue 2 or later approved revision.</td>
<td>- Avionics Mobile Services Project Certification Plan AM1498 Reference/PART2-003/149 8-01 or later approved revision - Avionics Mobile Services AFM Supplement reference AFM PART2-008 Issue 2 or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;, Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>10/10/2018</td>
<td>Active</td>
<td>10014198</td>
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<tr>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 1</td>
<td></td>
<td>767-300</td>
<td>Introduction of 279 passenger configurationRev. 1: Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>Aerospace Engineering Solutions Modification AES-707-026 dated April 13 2005 or later approved revision.</td>
<td></td>
<td></td>
<td>28/04/2021</td>
<td>Active</td>
<td>10014208</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 2</td>
<td></td>
<td>CESSNA 182-Q</td>
<td>Installation of Noise Reduction Kit C182-406650</td>
<td>- Anhang zum Flughandbuch SPOH, 12498_1050_HM</td>
<td>1. The approval is valid only in combination with the installation of FAA STC SA0091335 (Texas Skyways Inc.) or STC SA0015296 (Air Pains Services Corp.) Both with own EASA approval 2. The approval for the models Cessna 182 E, F and G is not valid for the combination with the installation of S...</td>
<td>03/02/2022</td>
<td>Active</td>
<td>10014214</td>
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Notes:
- STC Holder: The company or organization that issued the STC.
- Rev.: Revision level.
- Foreign/NAA: Foreign Aircraft Approval Authority or National Aircraft Approval Authority.
- Mfr.: Manufacturer.
- Description: A description of the STC.
- Certification basis: The certification basis provided by the EASA.
- Associated Technical Documentation: Reference to related documents.
- Limitation/Condition: Limitations or conditions associated with the STC.
- Date of Issue: The date the STC was issued.
- Status: Whether the STC is active, expired, or revoked.
- System Record: Unique identifier for the STC.
<table>
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<th>STC Holder</th>
<th>STC ref.</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 2</td>
<td>Avionics Mobile Services Modification AMI/478. Installation of a/t/home Collision Avoidance System (ACAS II).</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft type configurations shall be determined by the installer.</td>
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<td>27/02/2018</td>
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<td>LUTTHANSA TECHNOLOGY AG</td>
<td>REV. 1</td>
<td>Installation of Antenna, Receiver Transmit System, and Network services. This STC installs through the cabin a set of equipment to allow dataexchange. The Revision 1 has introduced some modifications through the initial STC design by upgrading the existing Wireless Access Points (WAPs) to support the IEC-61965.</td>
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<td>27/02/2013</td>
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<td>CEA</td>
<td>REV. 9</td>
<td>Installation of TAE 125 engine2. Installation of TAE 125-02-99 engine and installation of FADEC. BackupBattery3. Installation of TAE 125-02-1 14 engine/revision 9 follows implementation of DET 211005 Amld F dated 12/03/2024.</td>
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<td>15/03/2024</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 2</td>
<td>Extension of EASA STC EASA.A.S.01402 Installation of the 2-blades MTV-21-A770-05</td>
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<td>04/11/2010</td>
<td>Active</td>
<td>10014230</td>
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<td>AEROTECH CONCEPT</td>
<td>REV. 6</td>
<td>Cabin Baby Bassinet Installation Rev: 4. Revision of EASA Approved Model List (AML) to cover following ATC Type: A350, B787 &amp; EMB-190.</td>
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<td></td>
<td>31/01/2022</td>
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<td>10014248</td>
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<td>BEECHCRAFT BERLIN AVIATION GmbH</td>
<td>REV. 2</td>
<td>Installation of Cloud Research and Seeding System This EASA STC 10014250(-x.xx) REV. 2 reflects the transfer of STC EASA.A.S.01472 Revision 1 from Scandinavian Avionics Design GmbH to Beechcraft Berlin Aviation GmbH</td>
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<td>27/09/2010</td>
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<td>STC Holder</td>
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<td>AVIATION TRADERS LIMITED/EUROPE S.A.S.</td>
<td>10014254</td>
<td>(EASA.A.S.01479)</td>
<td>1</td>
<td></td>
<td>707-300</td>
<td>Avionics Upgrade Rev. 1: Transfer of STC</td>
<td>1. ATL Modification ATL/707M127.2-Compliance Plan CP10/295/07 - Issue 63. Flight Manual Supplement ATL/AFM Supplement Number 34 Issue 1.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>10014266</td>
<td>(EASA.A.S.01495)</td>
<td>4</td>
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<td>F177RG</td>
<td>Rev 4: Provisions for three attachments in the right side of the instrument panel in accordance with the STC EASA.A.S. 01496. Additional applicability to F177RG SIN 047</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval.</td>
<td>- Supplement Flight Manual Rev. 1 dated 16/11/2005 - later Rv.5 revisions of the above listed documents approved by EASA. - Troyes Aviation REF 51 - Minor Change Troyes Aviation REF 51 Revision 2 July 2006 - Minor Change Troyes Aviation REF 45 approved under EASA.A.S.01465/Revision 1</td>
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<td>CESINA DUSSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td>10014271</td>
<td>(EASA.A.S.01501)</td>
<td>1</td>
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<td>525, 525A</td>
<td>Installation of Honeywell KMH 880 or KMH 880</td>
<td>The Certification Basis for the original product and the following: Additional or alternative airworthiness requirements are applicable to this certification approval FAR-23 The requirements for environmental protection and the associated/electrified noise and/or emissions levels of the product are unchanged and remain applicable to this certification approval without any impacts.</td>
<td>Installation Procedure 2003-10-1 Rev. CF Flight Manual Supplement 2003-10-8 Rev. DMaintenance Manual Supplement 2003-10-9, Rev. 8</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>10014281</td>
<td>(EASA.A.S.01518)</td>
<td>3</td>
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<td>737 SERIES 737-300-300ER 737-378-300</td>
<td>Installation of non-electrical FPM System Rev. 1: product name updateRev. 2: applicability extended to the “Max” model 737-38-378-9 Rev. 3: types corrected</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated/electrified noise and/or emissions levels of the product are unchanged and remain applicable to this certification approval without any impacts.</td>
<td>Supplemental Type Definition Document STD23-05/03M Issue 2Rev.1 Airplane Flight Manual Supplement A-33-05/24-AM5. Approval of Continued Airworthiness ICA33-0503M Rev 2Master Minimum Equipment List Supplement A-00-19/88-MMEL Rev 2</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>10014282</td>
<td>(EASA.A.S.01519)</td>
<td>1</td>
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<td>BOEING 747-400</td>
<td>WVIP Cabin installation Rev. 1: Replacement of conductive oxygen flex - hoses with non-conductive ones. Renumbering the certificate from EASA.A.S.01519 to 10014282.</td>
<td>The Certification Basis for the original product as amended the following: additional or alternative airworthiness requirements - Special Condition (s): CRI D-1.1 “Isolated CRI D-1.3 &quot;Firm Handhold&quot; Issue 1...</td>
<td>Generic Documents STD2 25-00/60M Issue 2. Rev 2Cabin installation documents. STD2 25-00/19 Issue 1: AFM Supplement A-00-06/181-AMF Issue 1 rev.SICA document A-00-06/137 Issue 1 rev O2 Oxygen flex hoses replacements: CPL 35-20/57M Rev 0. Rev 1 later revisions of the above listed documents approved</td>
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<td>EASA STC no.</td>
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<td>10014287</td>
<td>CONTINENTAL AEROSPACE TECHNOLOGIES</td>
<td>REV. 15</td>
<td>LBA EMZ SA1295</td>
<td>F172L, F172M, F172N, F172P, F172F</td>
<td>Major Change to STC EASA.A.S.01527 Revision 5.1. Installation of engine TAE 125-01 in accordance with Installation Manual M4-20-01 and Type Design Definition Doc No. TDD-20-01-1.2. Installation of engine TAE 125-02-99 in accordance with Installation Manual M4-20-02 and Type Design Definition Doc No. 3...</td>
<td>The Certification Basis for the original product as amended by this STC allows additional or alternative airworthiness requirements - Special Condition (a): CRI A-00-Issue 7 (Doc 13, 2007) - Equivalent Safety Finding (a): CRI A-00-Issue 7 (Doc 13, 2007). The following paragraph (a) at a later amendment JA...</td>
<td>21/02/2020</td>
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<td>10014292</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>7137-700 IGW</td>
<td>Revision of EASA.A.S.01534 IR per Major Change 001002729-001. Operational 8.99 psi maximum cabin differential pressure (Lower Cabin Altitude, CA) Original STC Title: 8713-700 IGW VIP Cabin 19 Pax.</td>
<td>- Supplemental Type Definition Doc. STD000-0501M Issue 03, 0415 Model DR300/120, DR300/120A; DR300/140, DR300/140B; DR300/120C, DR300/120D, DR300/125 DR315, DR340, DR360, DR380, DR380/140, DR380/140R, DR380/325, DR380/325, DR380/108, DR380/100C</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this STC: - CRI D-01 (Distance between Passenger Compartment + CRI C-02 (Isolated Compartment) + CRI F-01 (Executive Power System) - TGL No.17 (Passenger 5)</td>
<td>12/03/2012</td>
<td>Active</td>
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<td>10014302</td>
<td>SAFRAN SEATS</td>
<td>REV. 1</td>
<td>767-300</td>
<td>Business Class Footstool Assemblies on British Airways Boeing 767-300</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. Therequirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Health Tech Modification document C973 &amp; associated DDPS(&gt;)(&gt;) Issue 1(&gt;)(&gt;) or later approved revision.</td>
<td>14/03/2019</td>
<td>Active</td>
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<td>10014303</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 3</td>
<td>DR400/2+2, DR400/4+4L, DR400/100</td>
<td>DR400/180</td>
<td>Installation/Installation of a modification kit to allow the operation with MOGAS a.e. EN228 to aircrafts models and variants as listed above. The installation permits the installation of the existing fuel and associated fuel production.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this STC: Approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>General: Master Document List MDOCL_203026_rev1_20130415Installation Instruction S_20130263_20130264_20130341Maintenance Manual Supplement SPF0H_8969_DR300 rev1_20130341Installation Instruction S_20130263_20130264_20130341Maintenance Manual Supplement - Supplemental Maintenance Manual MMS_2030326_rev1_20130341</td>
<td>03/02/2022</td>
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<td>10014303</td>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 2</td>
<td>PA-31T2 (CHEYENNE IXL)</td>
<td>Certification of FAA approved field modifications for the issue of anEASA C of A.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph. “Description” Compatibility with other aircraft engine configurations shall be determined by the installer.</td>
<td>27/06/2018</td>
<td>Active</td>
<td>1014303</td>
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<td>EASA STC no.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10014305</td>
<td>FOKKER SERVICES B.V.</td>
<td>REV. 2</td>
<td>STC-10014305</td>
<td>737 SERIES</td>
<td>Number 1 Cockpit Windshields: Addition of layers of polyurethane interlayer to the existing qualified designs to improve serviceability. Rev. 2: Transfer from GKN Aerospace to Fokker Services</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>17/02/2017</td>
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<td>10014312</td>
<td>TENENCIA Ltd.</td>
<td>REV. 1</td>
<td>STC-10014312</td>
<td>737 SERIES</td>
<td>Hawker Beechcraft Baron Steep Approach Landing Certification/This Certification at Revision 1 of the STC ref. EASA.A.S.01575 includes additional aeroplane models compared to the initial issue of the STC</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<td>10014320</td>
<td>MAP AIRCRAFT PART 21 AS</td>
<td>REV. 3</td>
<td>STC-10014320</td>
<td>737-800</td>
<td>Change of the cabin layout to Swedish National Air Medevac (SNAM) SAS Project number 737-2520-01 and SAS Project number 737-2520-02: Transfer of STC from Scandinavian Airlines System to SASTechnical Services AB: Revision 3: Transfer of STC from SAS Technical Services AB to MAP Aircraft Part 21...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>SAS Project number 737-2520-01-SAS Technical Order MTO-255165E-SAS and MTO-255491A-SAS Flight Manual Supplement for Boeing 737-800 Supplement No: SAS-737- SNAM, Project No. SAS-737-2520-01-SAS drawing “737-800 Air Medevac Cabin Configuration”, No. GE-737.10108AS project number 737-2523-01...</td>
<td>SAS Project Number 737-2520-01- s/n 28318, 28324, 28325, 28326, 28327, 30195, 30196, 30197, 30469, 30470, 30471, 32276, 32277, 32278. SAS Project Number 737-2520-02: s/n 28319, 28323, 28324, 28325, 28326, 28327, 28328, 30194, 30195,30196, 30197, 30467, 30468, 30469 and 30470,30471, 32276, 32277...</td>
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<td>10014322</td>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>REV. 2</td>
<td>STC-10014322</td>
<td>A300 B4-103, A300 B4-203</td>
<td>Introduction of Enhanced Mode S TransponderRev. 2: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Aviation Traders Limited Modification 9613(&gt;)-ATL AFM Supplement 35 Issue 3, SSR Mode S Elementary &amp; Enhanced Surveillance (Without LHAS-STC ST01146A Installed)&lt;(&gt;,&lt;)&gt;-Atl AFM Supplement 47 Issue 1, SSR Mode S Elementary Enhanced Surveillance(With LHAS-STC ST01146A Installed)&lt;(&gt;,&lt;)&gt;-Atl AFM Supplement 47 Issue 2 to the existing approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer. Prior to installation of this change/repair it must be determined...</td>
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<td>10014323</td>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>REV. 1</td>
<td>STC-10014323</td>
<td>727 SERIES</td>
<td>Introduction of Terrain Awareness &amp; Warning System (TAWS)Rev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Aviation Traders Limited Modification ATL727/M165 Issue 3-Aviation Traders AFM Supplement No 7278-1 to the existing approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>Prior to installation of this change/repair it must be determined the ICAO/interrelation between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>STC Holder</td>
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<td>LUIFTANSA TECHNICK AG</td>
<td>747-400, 747-400F</td>
<td>Installation of non-electrical Floor Path Marking System</td>
<td>STC Holder</td>
<td>Limitations/Conditions</td>
<td>03/03/2004</td>
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<td>10014352</td>
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<td>SAAB AB</td>
<td>LEARJET 35A 35A</td>
<td>Installation of Target Mission Operators Console</td>
<td>Revision 1: Transfer from Saab Nyge Aero AB to SAAB AB.</td>
<td>Saab Nyge Aero Technical Directive-Modification, TDM03-03.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other subsequently installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/12/2023</td>
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<td>10014331</td>
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<td>GAMA ENGINEERING Ltd.</td>
<td>200, 300</td>
<td>Installation of a Ground Use Battery (GUB) system Gama Aviation Mod rEHLM/ MOD1370.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>Installation in accordance with GAMA Aviation Modification HL/ MOD 1370 as defined by Master Drawing List (MDL) ref MDL-1370 Issue 1, dated 5 January 2015, and installation instructions IP-1370 Issue 1, dated 5 January 2015. Operation in accordance with GAMA Aviation Flight Manual Supplement ref/AFMS-1370...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other subsequently installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2015</td>
<td>Active</td>
<td>10014307</td>
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<tr>
<td>AVIATION TRADERS LIMITED</td>
<td>575-200</td>
<td>Introduction of Enhanced Mode S Transponder: Rev. 1. Transfer/STC</td>
<td>EASA.A.S.01622</td>
<td>Installation and in accordance with GAMA Aviation Modification WRT / STC 360 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other subsequently installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
<td>Active</td>
<td>10014309</td>
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<td>SEGELFLUG GRUPPE SIEMENS MÜNCHEN/E.V.</td>
<td>HK 36 TC</td>
<td>Optional operation with skis with directional control and breakingsystem instead of wheels in winter</td>
<td>EASA.A.S.01629</td>
<td>Englauung 1 zum Waartgungshandbuch, Version 2 vom 04.02.2012; Ergänzung 5 zum Flughandbuch, Version 2 vom 30.11.2012; later revisions of the above listed documents approved by EASA.</td>
<td>Limited to S/N 36730 fitted with Skis. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other subsequently installed design change/repair it must be determined that the interrelationship between this design change and any other subsequently installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/01/2014</td>
<td>Active</td>
<td>10014302</td>
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<td>CONTINENTAL AEROSPACE TECHNOLOGIES</td>
<td>PA-28-181 (ARROW II) PA-28-180 (ARROW I) PA-28-161 (WARIOR II) PA-28-160 (WARIOR I) PA-28-150 (CHEROKEE WARRIOR) PA-28-151 (CHEROKEE WARIOR) PA-28-140 (CHEROKEE CRUISER)</td>
<td>1. Installation of engine TAE 125-01 in accordance with Installation Manual IM-40/01 and Type Design Definition Doc. No.: TDD 40-01.2.</td>
<td>EASA.A.S.01578</td>
<td>1. This STC superseded LBA-EMZ SA 1377 and former EASA STC A.3.01632.2 Power/Rotational Speed for TAE 125-01 and 125-02-99 installations: Take-off power: 99 kW at 2300 min-1; Propeller speed: 99 kW at 2300 min-1 (Propeller speed); for TAE 125-02-114 installations: Take-off power: 99 kW at 2300 min-1; Propeller speed:</td>
<td>12/08/2019</td>
<td>Active</td>
<td>10014364</td>
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**Data record**

03/07/2024

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<td>365 AEROSPACE DDS LIMITED</td>
<td>MOD/757/1272</td>
<td>-</td>
<td>Boeing 757-200</td>
<td>Installation of an Enhanced Surveillance Mode S System</td>
<td>Aerospace Design &amp; Engineering Consultant/AFM Supplement 1 FMA-0014-001 Issue Aerospace Design &amp; Engineering Consultant/Flight Manual Supplement FMA- 1576-083 Revision for later revisions of the above listed document(s) approved/ accepted under the EASA system Aerospace Design &amp; Engineering Consultant.</td>
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<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/08/2019</td>
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<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>MOD/757/1272</td>
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<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>TCAS II System - Litton LTN-72R</td>
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<td>GAMA ENGINEERING LTD.</td>
<td>REV. 1</td>
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<td>BAE.125 SERIES 800</td>
<td>Title: Upgrade of the dual Flight Management System (FMS). Description of change: The original dual Honeywell FMZ-400 FMS (supplag to a dual Honeywell FMZ-2000 FMS.</td>
<td>CS-25 The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installed in accordance with Gama Engineering Ltd modification NLMD/889, reference the Master Documents List (MDL), document/references MDL-689-01, Issue 1. Other technical documentation: Operation in accordance with Aircraft Flight Manual Supplement, document/references AFMS89/901, Issue 2 Mandatory.</td>
<td>The installer must determine whether this design change is compatible with previously approved modifications.</td>
<td>05/10/2012</td>
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<td>10014392</td>
<td>TÄBY ARMMAINTENANCE EKTSIBOLAG</td>
<td>REV. 3</td>
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<td>SAAB SF430A</td>
<td>Saab 340 Class E Cargo Compartment as defined in document ST34MOD044</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated/noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- S21 FMS-0013 Issue 2 Flight Manual Supplement. - S21 MDEL-0005 Issue 1 Master Minimum Equipment List Supplement. - S21TREP-00 04 Issue 1 Mass and Balance Report. - S21 TEC-027 Supplement Manual Issue 4 - Instructions for Continued Airworthiness - Maintenance Programme.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>17/02/2020</td>
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<td>10014398</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
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<td>BOEING 747-400</td>
<td>&quot;Bahrian Royal Flight&quot; VIP cabin installation: Rev 1 Replacement of conductive oxygen flex-hoses with non-conductive ones. Certificate renumbering from EASA A.S.01692 Rev. 2. Type corrected</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated/noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Generic documents STDD 06-0317M Issue 02 Rev 02 VIP Cab installation documents AFM supplement A-00-0519-APR rev 02Instructions for continued airworthiness ICA-0-0317 Issue Oxygen hoses replacement documents CPL-35-2006M Rev 0r later revisions of the above listed document(s) is approved/accepted.</td>
<td>As per AFM supplement A-00-0519-APR Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/01/2021</td>
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<td>10014403</td>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 1</td>
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<td>SR Technica (it) documents:ST73-2S-1024- CCS, Rev 3</td>
<td>Reconfiguration of six Boeing 737-33V from 149TC to 16BC/100TC including installation of G2, G7 and LAV. Rev 1: Transfer of STC from SR Technica Switzerland to gROWING Engineering Partners GmbH</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>20/04/2021</td>
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<td>10014407</td>
<td>365 AEROSPACE DDS LIMITED</td>
<td>REV. 2</td>
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<td>737-200-300-400-500-600</td>
<td>Installation of Dual High Frequency (HF) Communications System: Rev 2 Transfer of STC from 365 Aerospace Ltd. to 365 Aerospace DDS Ltd.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated/noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.</td>
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<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>REV. 1</td>
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<td>A300F4-608ST</td>
<td>Installation of ISS by A300-600/605 installation of two Integrated Stand-by Instrument System (ISS) in replacement of the stand by speed, attitude and altitude.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified/noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Change Approval Sheet, ref: ST34MOD015 issue 1AFM Supplement ref: ST34MOD015 dated 19 October 2006Change Certification/Amendment ST34MOD044, CCS Issue 3 dated 07.02.2016 last amendments of the above listed documents approved by EASA.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or voided.</td>
<td>11/07/2016</td>
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<td>10014409</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>REV. 1</td>
<td>CESSNA 650</td>
<td>Upgrade of two existing Mode S transponders. Rockwell Collins(CRD)-94D by Rockwell Collins SB 501 Replacement of Mode S Controller CTL-52 by CTL-92E along with installation of new wiring for the associated certified noise and/or emissions levels of the original product.</td>
<td>Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document Ltd 163A-MDL, Rev. A from 27 October 2010/Appendix Flight Manual Supplement 163A-AFM, Rev. A from 28 October 2010 later revisions of the above listed documents approved by EASA.</td>
<td>Aircraft Serial Numbers 650-7001 through 650-7119</td>
<td>29/10/2010</td>
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<td>10014411</td>
<td>TUI AIRLINES BELGIUM N.V.</td>
<td>REV. 1</td>
<td>767-300</td>
<td>Introduction of BE Aerospace Spectrum Seat in accordance with Thomson/Indret Modification Summary MS/7871462 Issue 3. History: The STC certificate A.S. 01705 is re-issued because Thomson/Indret (previously known as Thomson/Indret) part of the TUI Group no.781462 transfers the STC to another TUI...</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Thomson/Indret Modification MOD/ 7671462 Issue for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>23/12/2016</td>
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<tr>
<td>10014417</td>
<td>TÁBY ARMMAINTENANCE &amp; ACTIEBELAS</td>
<td>REV. 3</td>
<td>SAAB SF340A</td>
<td>Introduction of Alternate Cargo Net Location Lay-out 1: to add an additional configuration that extends bay lengths by utilising nets installed in any of the eight positions.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EASA Approved - STC Twenty One Ltd Flight Manual Supplement No. S2.1-FMS-0014 Issue 6 - STC Twenty One Ltd Supplement Manual - ICA No. S21.TEC-0116 Issue 8; EASA Accepted - STC Twenty One Ltd Master Document List No. S21. MDL-1508 Issue 2.</td>
<td>Prior to installation of these changes/repair it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/02/2020</td>
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<td>10014422</td>
<td>RUAG AG</td>
<td>REV. 2</td>
<td>PILATUS PC-12/47</td>
<td>Installation of Dual Garmin GNS-530 COMNAV/GPS System. (Note: This STC is the Revision 2 of EASA STC 2004-4674 dated 03/02/2004).</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Engineering Technical Order TSA PC-12/34.60.02EO Revision Car later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>01/02/2021</td>
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<td>10014423</td>
<td>ISRAELI AEROSPACE INDUSTRIES LTD IAIA</td>
<td>REV. 5</td>
<td>CAA ISRAEL STC SA152</td>
<td>Conversion from Passenger to Special Freighter Configuration. Revision 1: Boeing 747-400SF conversion from PAX to Freighter CAAI STCs A3673 &amp; A3674.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Israel Civil Aviation Authority STC SA152 dated October 27, 2006(x&lt;-&gt;x). MDL 9W-MDL-905-00-05- A3673 Revision AH(x&lt;-&gt;x); AIM-S 1: IAI Doc No. 385-5001-A3663-A361 revision 1, October 24, 2012-Supplement to Airplane Flight Manual D86/1002-81 for IAI B747-400SF with GE engines(x&lt;-&gt;x); AIM-S 2: IAI Doc.</td>
<td>Refer to the limitation section of the applicable AFM Supplement for approved/accepted at the time of this STC, depending on the serial number of the aircraft for installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>08/04/2019</td>
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<td>10014426</td>
<td>ROEDER PRECISION GMBH</td>
<td>REV. 1</td>
<td>441</td>
<td>Installation of Dual Mode S Transponder GTX 330D</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Engineering Order C441-NY-0124-4 AFM &amp; C441- NY-0124-08, Rev. 0, approved 14 February 2006, or latest EASA approved revision - Maintenance Procedures C441-NY-0124-07, Rev. 0, or later approved revision.</td>
<td>1. Limitation according AFM-S C441-NY-0124-4 and C441-NY-0124-08, approved under EASA STC 10043502(formerly EASA A.S. 01681). Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>25/09/2017</td>
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<td>10014431</td>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 2</td>
<td>1124A</td>
<td>Installation of Airborne Collision Avoidance System (ACAS II).</td>
<td>Autonics Mobile Services Modification AM 1488 Autonics Mobile Services AFM Supplement No. PART21-0098/017 Revision - dated Apr 08 2005 or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>27/06/2018</td>
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<td>10014435</td>
<td>ROEDER PRAEVISION GMBH</td>
<td>REV. 2</td>
<td>HAWKER 800/-800XP 850A/-800B BAE-125 SERIES 850A/-800B BAE-125 SERIES 1000A/-1000B</td>
<td>Installation of wing mounted Xenon Taxi' Landing Light and/or Radommounted Taxi Light</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Engineering Order HB00- NY-0126-05- AFM-S HB00- NY-0125-08 Rev. 0 approved 07/04/2006 Maintenance Procedures HB00-NY-0125-07, Rev. 0, later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>13/02/2018</td>
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<td>ROEDER PRAEVISION GMBH</td>
<td>REV. 1</td>
<td>HAWKER 800/-800XP 850A/-800B BAE-125 SERIES 1000A/-1000B</td>
<td>Installation of Hidrum Satcom C1000</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Engineering Order HB00- NY-0126-01- AFM-S HB00- NY-0128-08, Rev. 0, approved 12. May 2006 Maintenance Procedures HB00-NY-0125-07, Rev. 0, later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>13/02/2018</td>
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<td>10014442</td>
<td>THEILERT AIRCRAFT ENGINES GmbH</td>
<td>REV. 1</td>
<td>SR22</td>
<td>Installation of engine Centurion 4.0 in accordance to Type Design Definition Doc. No.: TDD-70-01/Engine: Centurion 4.0/Propeller: MTU-14/2 195-390/Certification Basis: see CRI A.1-max. Take Off Power: 228 kw at 2300 min-1 (Propeller-RPM)/max. Continuous Power: 228 kw at 2300 min-1 (Propell...</td>
<td>Supplement Pilot's Operating Handbook for the Cirrus SR 22 equipped with TAE Centurion 4.0 installation: Doc. No.: 70-0010-70021, Issue 1 or later approved issue Maintenance Manual Cirrus SR 22 / Centurion 4.0, Doc.: AMM-70-01- Operation &amp; Maintenance Manual Centurion 4.0...</td>
<td>19/06/2013</td>
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<td>10014445</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 3</td>
<td>P.68-TC “OBSERVER”, P.68-TC “VICTOR”, P.68-TC “OBSERVER”, P.68-TC “VICTOR”, P.68-TC “OBSERVER”, P.68-TC “VICTOR”, P.68-TC “OBSERVER”, P.68-TC “VICTOR”</td>
<td>Installation of 3-blade constant speed propeller MTU-12-B-C-F7/CF183-59b with MT-Propeller AFM Supplement AMM-70-01-70021, Issue 1 or later approved issue Maintenance Manual Cirrus SR 22 / Centurion 4.0, Doc.: AMM-70-01- Operation &amp; Maintenance Manual Centurion 4.0...</td>
<td>According to MT-Propeller Master Documentation List No. E-2783, Rev. 3, To be installed; According to MT-Propeller AFM-S Document No.E-1370, Rev. 2 when the MTV-12-B-C-F7/CF183-59b has been installed; According to MT-Propeller Master Documentation List No. E-2783, Rev. 3, To be installed.</td>
<td>According to MT-Propeller Master Documentation List No. E-2783, Rev. 3, To be installed in accordance with MT-Propeller AFM-S Document No.E-1370, Rev. 2 when the MTV-12-B-C-F7/CF183-59b has been installed.</td>
<td>30/11/2017</td>
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### List of EASA Supplemental Type Certificates

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<td>365 AEROSPACE DDS LIMITED</td>
<td>10014460</td>
<td>HS.125 SERIES 700A/B</td>
<td>10014481</td>
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<td>328 SUPPORT SERVICES GmbH</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
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<td>PIPER PA-23-250</td>
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<td>SAAB AB</td>
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<td>BEECH A100</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>10014490</td>
<td>MODEL 36, MODEL 36A, MODEL 35, MODEL 35A</td>
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#### Description
- **Installation of an Enhanced Surveillance Mode S System / Installation of new software versions of the Enhanced Surveillance Mode S System**
- **Installation of a Digital Air Data Display System**
- **Installation of a Digital Air Data Display System**
- **Modification of Piper PA-23-250 aircraft**
- **Modification of SaaB A100 aircraft**
- **Installation of Dual Universal Uns-1E FMS**
- **Modification of the SaaB A100 aircraft**

#### Certification Basis
- **CS25, AMDt. 01**
- **CS25, AMDt. 01**
- **CS25, AMDt. 01**
- **CS25, AMDt. 01**
- **CS25, AMDt. 01**
- **CS25, AMDt. 01**

#### Certification Program
- **13452-REV0**
- **13452-REV0**
- **13452-REV0**
- **13452-REV0**
- **13452-REV0**
- **13452-REV0**

#### Date of Issue
- **03/07/2024**
- **27/09/2018**
- **05/08/2019**
- **27/09/2018**
- **07/12/2023**
- **07/05/2013**

#### Status
- **Active**
- **Active**
- **Active**
- **Active**
- **Active**
- **Active**

#### System Record
- **10014452**
- **10014452**
- **10014452**
- **10014452**
- **10014452**
- **10014452**

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**Notes:**
- This STC is approved only for the product configuration as defined in the approved design data referenced in the paragraph “Description”. Compatibility with other aircraft configuration changes shall be determined by the installer. Prior to installation of this design change it must be determined that...
- This STC is approved only for the product configuration as defined in the approved design data referenced in the paragraph “Description”. Compatibility with other aircraft configuration changes shall be determined by the installer.
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<tr>
<td>10014495</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 1</td>
<td>767-300</td>
<td>Introduction of 200 passenger place interior configurationRef 1: Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>Aerospatiale Solutions Modification AES-761-492</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/04/2021</td>
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<td>10014496</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 2</td>
<td>P.68B &quot;VICTOR&quot;, P.68C/-</td>
<td>Camera Port Installation Rev. 2: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>None</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
<td>Active</td>
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<td>10014505</td>
<td>DASSAULT FALCON SERVICE</td>
<td>REV. 1</td>
<td>MF900 &amp; F900EX</td>
<td>200 VAC ELECTRICAL DISTRIBUTION ON AIRCRAFT STC EASA.A.S.02575 revision for applicability extension and addition revision 2.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable (toth certificate/approval)§25.859 (a) (1)(d4) at JAR 25 Change 14. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable...</td>
<td>None</td>
<td>13/10/2010</td>
<td>Active</td>
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<td>10014507</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 3</td>
<td>A330 SERIES A330-341/341 A330-341/342-343 A330-321/320-323 A330-321/320-303 A330-522/343 A330-201/202-203</td>
<td>Installation of non-electrical Power Path Marking System Certification 1: GuideU series 300408/09/04051000; all models except A330-841 and A330-941- Configuration 2: GuideU series 1000; all modelsRevision 3: Extended applicability (A330-841 and A330-941 models)</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) is added: “after an annoucement: For Configuration 1: EASA CS 25.029 (a), 25.301(a)(b), 25.303, 25.305 (a)(b), 25.307(a)(b)25.561(a)(b) (d), 25.8...</td>
<td>None</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/11/2023</td>
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<td>10014508</td>
<td>SAAB AB</td>
<td>REV. 1</td>
<td>LEARJET 35A</td>
<td>Mission PowerRev 1: Transfer from SAAB Aerotech AB to SAAB AB.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>SAAB Aerotech Modification TDM06-01-1-CP</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/12/2023</td>
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<td>EASA STC No.</td>
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<td>Rev. Level</td>
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<td>Models</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10014519</td>
<td>SAFRAN SEATS</td>
<td>REV. 2</td>
<td></td>
<td>747-400</td>
<td>Cathay Pacific B747 Cabin Re-Configuration</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remain applicable to this certificate/approval.</td>
<td>Modification CI030 Revision 1</td>
<td>1. Restricted to MSN 27332, MSN 27069 &amp; MSN 265522 Zero occupancy. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/02/2019</td>
<td>Active</td>
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<td>10014527</td>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td></td>
<td>CL-600-2D24 (RJ SERIES 900) CL-600-2C10 (RJ 700.701.702) CL-600-2B10 (RJ SERIES 100)</td>
<td>Installation of RAAS compatible EGPWS Computer in CL-600-2B10/-2C10/-2D24 Series Aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remain applicable to this certificate/approval.</td>
<td>- Document Inventory List CRJ-NY-0429 Revision 1, dated 09.02.2007 - Flight Manual Supplement CRJ- NY-0429-08 Revision 00r later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/02/2018</td>
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<td>10014528</td>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td></td>
<td>AVRO 146-RJ85 AVRO 146-RJ100</td>
<td>Activation of RAAS function of EGPWS computer in AVRO-RJ85/AVRO-RJ100 Series Aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remain applicable to this certificate/approval.</td>
<td>AFM - RJ85-NY-0428-08 Rev 0 - Master Document List RJ85-NY-0428-Rev 0</td>
<td>1. As listed in the AFM 2. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2017</td>
<td>Active</td>
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<td>10014537</td>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 2</td>
<td></td>
<td>RJ85/100</td>
<td>Installation of RAAS compatible EGPWS Computer in AVRO RJ85/100 Series Aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remain applicable to this certificate/approval.</td>
<td>AFM - RJ85-NY-0427-08 Rev 0 - Master Document List RJ85-NY-0427-08 Rev 0r later revisions of the above listed documents approved by EASA.</td>
<td>As listed in the AFM - Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/10/2017</td>
<td>Active</td>
<td>10014537</td>
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<td>10014538</td>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td></td>
<td>CL-600-2D24 (RJ SERIES 900) CL-600-2C10 (RJ 700.701.702) CL-600-2B10 (RJ SERIES 100)</td>
<td>Activation of RAAS function of EGPWS Computer in CL-600-2B10/-2C10/-2D24 Series Aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remain applicable to this certificate/approval.</td>
<td>- Document Inventory List CRJ-NY-0432 Revision 1, dated 09.02.2007 - Flight Manual Supplement CRJ- NY-0432-08 Revision 00r later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/02/2018</td>
<td>Active</td>
<td>10014538</td>
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<td>10014542</td>
<td>GAMA ENGINEERING Ltd.</td>
<td>REV. 1</td>
<td></td>
<td>A340-311, A340-312, A340-313</td>
<td>Child Restraint System - CRS - Installation in A340-300</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval: CRI F-11, “Child Restraint System”. The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Certification Plan, ref. doc. No. TN1785-004, Rev H, dated 15th December 2006 - Certification Compliance Check List, ref. doc. No. TN1785-005, Rev E, dated 8th June 2007 - Declaration of Design and Performance, ref. doc. No. DDP-1785-026, Rev. H, dated 20th August 2007 or later revisions of the above.</td>
<td>Limitations in accordance with § 4 - Limitations of the above mentioned DDP, ref. No. DDP-1785-026 All cabin interiors must be evaluated for compatibility of the Child Restraint System (CRS). The CRS is not approved for aeroplanes that are not defined by § 4 of the above referenced DDP.</td>
<td>10/12/2012</td>
<td>Active</td>
<td>10014542</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>GAMA ENGINEERING LTD</td>
<td>(EASA.A.S.02630)</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A340-643</td>
<td>Child Restraint System - CRS - Installation in A340-600</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CRS-F.01, “Child Restraint System”. The requirements for environmental protection and the associated certificated noise and/or emissions remain applicable to this certificate/approval.</td>
<td>- Certification basis: ref. doc. No. TN785-004, Rev. H, dated 10/5/2014. Certification Compliance Check List, ref. doc. No. TN785-006, Rev. E, dated 8th June 2007. Declaration of Design and Performance, ref. doc. No. DDP-1785-026, Rev. H, dated 22nd August 2007 or later revisions of the above.</td>
<td>Limitations in accordance with § 4 - Limitations of the above mentioned DDP, ref. No. DDP-1785-026. All cabin interiors must be evaluated for compatibility of the Child Restraint System (CRS). The CRS is not approved for airframes that are not defined by § 4 of the above referenced DDP.</td>
<td>12/10/2012</td>
<td>Active</td>
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<tr>
<td>GAMA ENGINEERING LTD</td>
<td>(EASA.A.S.02631)</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B747-400</td>
<td>Installation of MAGE Child Restraint System as detailed in Modification Leaflet for Modification no. B744-1795.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>As defined in Marn Aviation Group (Engineering) Ltd. (MAGE) Modification Leaflet for Modification No. B744-1795 or later revisions of the above listed documents approved by EASA</td>
<td>20/06/2012</td>
<td>Active</td>
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<td>DOA AESGLOBAL LIMITED</td>
<td>(EASA.A.S.02655)</td>
<td>REV. 1</td>
<td></td>
<td>767-300</td>
<td>Introduction of MainAPU Battery Paralleling for Extended Standby Power Time Capability Rev. 1 Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>Aerospace Engineering Solutions Modification AES-767-659</td>
<td>none</td>
<td>28/04/2021</td>
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<td>10014557</td>
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<td>SCAI TECH (SOULDLERE CHAUDRONNIER E ALU (NOX))</td>
<td>(EASA.A.S.02657)</td>
<td>REV. 5</td>
<td></td>
<td>DR 400/180 B, DR 400/180 D, DR 400/180 B, DR 400/160, DR 400/160 A, DR 400/120, DR 400/120 A, DR 400/120 B, DR 400/120 C, DR 400/120 D, DR 400/120 E</td>
<td>SCAI-DR319, DR300-108, 120, 125 (DR300-108, 120, 125), DR320, DR320, DR320, DR320</td>
<td>Suplement au manuel de maintenance SCAI-DR315, DR300-108, 120, 125 (DR300-108, 120, 125), DR320, DR320, DR320</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 2.A.109. Prior to installation of this design change must be determined that the interrelationship between this design change and any or previously installed design change and/or repair will introduce no adverse effect upon the a...</td>
<td>29/10/2020</td>
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<td>LUFTHansa TECHNik AG</td>
<td>(EASA.A.S.02663)</td>
<td>REV. 2</td>
<td></td>
<td>A300 SERIES, A310 SERIES</td>
<td>Installation of non-electrical Floor Path Marking System</td>
<td>1. Supplemental Type Definition Document STD03-0602M, issue 01 dated 16.05.2002. Airplane Flight Manual Supplement A-33-0624-AFM3. Applicant’s Declaration of Compliance dated 29.06.2006 or later revisions of the above listed documents approved by EASA</td>
<td>05/03/2014</td>
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<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>(EASA.A.S.02672)</td>
<td>REV. 2</td>
<td></td>
<td>A319-112</td>
<td>In-flight telecommunication equipment – antenna installation/partial removal of GTE Airline System installed by FAA STC/STC/1329194Y-Owned SA1695G. The Revision 1 was issued to reflect the transfer from SR Technics AG to SR Technics Switzerland LTD. Rev. 2 Transfer of STC from SR Technics Switzerland...</td>
<td>GTE Airline STC Documentation Checklist AFD/00016-00, Rev.03. Technica Doc. C3320-23-100S-CSS-01 Rev.08 or later revisions of the above listed documents approved by EASA</td>
<td>none</td>
<td>20/04/2021</td>
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<tr>
<td>10014593</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 1</td>
<td>HK 36 TTC-ECO</td>
<td>HK 36 (RESTRICTED)</td>
<td>Modification of the Dimona HK 36 R &quot;Super Dimona&quot; equipped with the engine Rotax 912 A3 with the alternative Propeller MTV-21-A-CF/CI75-05 Rev. 1 Transfer to ACC Columbia Jet Service GmbH</td>
<td>Service Bulletin GMS SSB 2008-2/ ziehung zum Flugerbuch für den Motorsegler HK 36 R SPOH, 14326_rev_23082 037 Ausgabe 08.2007 Wurtungsanhangbuch für den Motorsegler HK 36 R. 3024543-HK36-WLB. Ausgabe 25. Juni 1990/umfangenweisung, Modification 14328</td>
<td>All S/N of the variant HK 36-R &quot;Super Dimona/ Prior to installation of this change/repair it must be determined thatthe interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>15/01/2022</td>
<td>Active</td>
<td>1011632</td>
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<td>10014593</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>BOEING 737-700 IGW</td>
<td>BOEING 737-700</td>
<td>Upgrade of existing VIP cabin configuration from private operations to commercial operations. Rev. 1: Replacement of conductive oxygen flex hoses with non-conductive ones. Renumbering certificate from EASA A.S.0705 to 10014593.</td>
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<td>10014594</td>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>REV. 1</td>
<td>BAE 146 SERIES</td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: As defined in OAR A-100 Issue 2 dated June 11th 2001 The requirements for environmental protection and the associated certified noise and/or emissions levels of the product.</td>
<td>Generic documents STD DOO 01_01_00 M Issue 07 rev 03 For the change to commercial operations CP 05_06_17M Issue 03 CR 05_06_17M Issue 01 STO 25_06_17M Issue 01 rev 01 03A 00_01_31_11LM supplement Cabin TT&lt; (&gt;=&lt;)=L Issue 03A 05_06_17M AFM supplement Cabin TT&lt; (&gt;=&lt;)=L Issue IRA 05_01_31_11LM supplement Rev 12F.</td>
<td>- Limitations as per AFM supplements A 00_01_31 and A 25_06_17M. The Luftfahrt Bundesamt STC LBA. 21E2.TA0307 is a prerequisite for this installation. Prior to installation of this change/repair it must be determined thatthe interrelationship between this change/repair and any other previously installed...</td>
<td>17/11/2020</td>
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<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>REV. 1</td>
<td>DC 8-60</td>
<td></td>
<td>The Type and OSD Certification Bases (CB) for the original product are applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w.</td>
<td>包装 Airplane Flight Manual Supplement No. 44 Issue 1, dated 7th September 2006 ATL Definition Document No. CD03840001 dated 7th September 2006 or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined thatthe interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
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<td>10014596</td>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>REV. 1</td>
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<td>Installation of Honeywell MKV7 EGPSWS Rev. 1: Transfer of STC</td>
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<td>10014598</td>
<td>B2000CIT, B200CIT, B200T, A200C, A200CT, B200, B200C</td>
<td>REV. 4</td>
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<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23 amendment 0.0 unless otherwise noted CS 23.301(a), 23.303, 23.305(a)(g), 23.307(a), 23.333(a), 23.603A, 23.605(a), 23.607, 23.609(a), 23.613, 23.619, 25.625(a), 23.785.</td>
<td>- Air Alpha design package EO-2000-26-105 - Flight Manual Supplement: FMS/EO-2000-26-105-Uniffs 3000 Installation and Maintenance Manual version 1 Aug 2006 Uniffs 3000 Operator Manual version 1, Aug 2006 Supplemental Installation and Maintenance Manual for King Air 200: Air Alpha N. ATP/0109/016</td>
<td>1. Compatibility of this design change with other previously approved/modifications must be determined by the installer. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation...&quot;</td>
<td>13/03/2019</td>
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certifications.basis: 10014604
STC Holder: ROEDER PRAEZISION GMBH
STC Number: DC-3C-R1830-90C
Model: DC-3C
Revision: REV. 2
Description: Installation of emergency exit signs and floor escape path markings at DC3C series aircraft installation of emergency exit signs light with power source and installation of fluorescent floor path markings
Certification basis: The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Certified Technical Documentation:
- Master Document List DC-3C-NY-0532 Rev. 0 dated 20 Nov 2008 for later approved revision
- Drawings DCS-3C-NY-0532-05 Rev 0. Mechanical Installation DC3- NY-0532-05 Rev. 0, Electrical Drawings DC3- NY-0531-02 Rev. 0, Weight & Balance DC3- NY-0532-09 Rev. 0

Limitations/Conditions:
The system must be maintained in accordance with Maintenance Procedures DC-3C- NY-0532-07 Rev. 0 dated 20 Nov 2008 or later approved revision. The system must be operated in accordance with the AFM supplement DC3- NY-0532-08 Rev. 0 dated 20 Nov 2008 or later approved revision. Prior to installation of this device...

Date of Issue: 08/10/2018
Status: Active
System Record: 10114602

10014607
STC Holder: FOKKER TECHNIEK B.V.
STC Number: CL-600-2C10 (RJ 700,701,702)
Model: CL-600-2C10
Revision: REV. 3
Description: VIP conversion of Bombardier CL600-2C10 (RJ-700) for Consolidated Contractors Company. as detailed in Fokker Services Engineering Report FS75-NSIA. Description of the modification Revision 2. Wireless handsets in the Aircraft telephone installation in aRJ7-700 for Consolidated Contractors Company (provi...
Certification basis: The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Certified Technical Documentation:
- If the utilization of the VIP converted aircraft deviates from the assumed utilization (e.g. flight hours or flight profiles) supporting the specified maintenance requirements as defined by Bombardier, the operator of the aircraft will have to review the adequacy of the maintenance tasks with the airworthiness authorities.

Date of Issue: 05/12/2019
Status: Active
System Record: 10114607

10014612
STC Holder: ROEDER PRAEZISION GMBH
STC Number: 200, 390, 1900
Model: Cabin Modification - Galaxy installation in the LH FWD Entrance Area - High Cabinet installation in the FWD Rh Cabin
Revision: REV. 1
Description: Installation of HD Nose Gear Lights.
Certification basis: The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Certified Technical Documentation:
- Installation to be in accordance with Nayak Engineering Order netBE200/300-NY-0733-05, Rev 0 dated 12 September 2006 or later original revision.
- Instructions for Continued Airworthiness as defined in Nayak Maintenance Procedures REF BE200/300- NY-0733-07, Rev 0 dated 12/September 2006, or later appr...

Date of Issue: 30/10/2017
Status: Active
System Record: 10114612

10014613
STC Holder: JET AVIATION AG
STC Number: S/N 32774, PA-4421 737-700GW (737-7EJ),
Model: Cabin Modification - Galaxy installation in the LH FWD Entrance Area - High Cabinet installation in the FWD Rh Cabin
Revision: REV. 2
Description: The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Certified Technical Documentation:
- AFMS BPKA200/201-01:Bar later revisions of the above listed documents approved by EASA.
- Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Date of Issue: 29/07/2016
Status: Active
System Record: 10114613

10014618
STC Holder: AVIATION ENGINEERING AS
STC Number: B200
Model: B200
Revision: REV. 2
Description: Antennas - Relocation and Installation in accordance with EO-B200-25-106 and D2W. No. EN-B200-25-106-01 to support UNIFIB 3000 Flight Inspection System.
Certification basis: The Certification Basis for the original product as amended by thefollowing additional or alternative airworthiness requirements: CI 23 amendment 23-0, 23S23.35(b), 23.350, 23.353,23.357, 23.359(b), 23.377,23.385, 23.571(c), 23.603, 23.605, 23.609, 23.611, 23.613, 23.627, 23.1525, 23.1529, 23 App GT...

Certified Technical Documentation:
- Flight Manual supplement FMS-B200/300-25-106, Rev.0, 07 June 06; Supplemental Maintenance Manual SM5-MPS-250104, Rev.0, 07 June 06; later revisions of the above listed documents approved by EASA.
- Prior to installation of this design change it must be determined that the interactions between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Date of Issue: 13/03/2019
Status: Active
System Record: 10114618

10014620
STC Holder: LUFTHANSA TECHNIK AG
STC Number: B737-700 IGW
Revision: REV. 5
Description: Revision of EASA.A.S.02752 Rev. 4 per Major Change 00120010/01. Operation at 8.99 psi maximum cabin differential pressure (Lower Cabin).Altitude LCA/Original STC Title: Design and implementation of VIP interior and unائرceilance aircraft systems.
Certification basis: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CRI D-01-07/2012: AFM Supplement Doc. A-00-07/784-ARM, Revision 10, dated 21/11.

Certified Technical Documentation:
- Instructions for Continued Airworthiness (ICA) Doc. A-00-07/784-ICA Issue 2
- Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Date of Issue: 03/12/2012
Status: Active
System Record: 10114620
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<th>EASA STC no.</th>
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<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
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<th>Limitations/Conditions</th>
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<td>10014635</td>
<td>AVIATION TRADERS LIMITED/EUROPE S.A.S.</td>
<td>REV. 1</td>
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<td>BAE 146 SERIES</td>
<td>Installation of active AirData Signal GNS-XLS Flight Management System</td>
<td>Rev. 1: Transfer of STC</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 1</td>
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<td>CESSNA 172S</td>
<td>Installation of Noise Reduction Kit</td>
<td>Rev. 1: Transfer to ACC Columbia Jet Service GmbH</td>
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<td>JET AVIATION AG, BASEL</td>
<td>REV. 1</td>
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<td>MF900</td>
<td>Wireless Access Service Point (WASP) Wireless LAN (WLAN)</td>
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<td>- Anhang zum Flughandbuch Lärmminderungskit C172S-806550- Umrüstanweisung 14768</td>
<td>None</td>
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<td>10014659</td>
<td>JET AVIATION AG</td>
<td>REV. 2</td>
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<td>737-700-800-900-900ER</td>
<td>Wireless LAN (WLAN) Activation/Expanding demonstration of the existing immunity to IEEE 802.11 ad/β/ α from the B737-700 to the B737-800/900/900ER and expand the immunity toIEEE 802.11 ac.</td>
<td>Master Data List, Doc. No. FIC200103-001, Rev. B, dated 04.04.2014- AFMS (SN 106), Doc. No. B3230403-001-13, Rev. 9, Date of Approval -ICA (SN 106), Doc. No. B3230403-001-13, Rev. 9, dated 05.05.2009- AFMS (SN 106 &amp; 004), Doc. No. JBSL-M-00122AFM-01, REV. 1, Date of approval - ICA (S/N 934 &amp; 104)</td>
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<td>REV. 2</td>
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<td>737-300, 737-400, 737-500</td>
<td>Mode ‘S’ Transponder System for Elementary and Enhanced Surveillance</td>
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<td>- Master Data List, Doc. No. BB3732261-001, Rev. D, dated 08. May 2014- AFMS, Doc. NO. JBSL-M-00141AFM-01, Rev. 1 for later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>H4 AEROSPACE (UK) Ltd.</td>
<td>REV. 3</td>
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<td>757-200</td>
<td>Cabin Reconfiguration on Boeing 757-200 aircraft</td>
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<td>BBJ Design Services Ltd Modification BBJ3607</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>Limitations/Conditions</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
<td>172RG</td>
<td>Modification 14839 INSTALLATION OF LOW NOISE SILencer KIT &quot;System Gomolzig&quot; CT172RG 606520 Additional Propeller MTV</td>
<td>Low Noise System Kit CT172RG 606520 Master Document List MDOCL 14839 rev 2 2013</td>
<td>Boeing 747-100B/-200F/-300F/-400F/-100/-300/-400/-100/070/-400/070</td>
<td>Boeing 747-100B/-200F/-300F/-400F/-100/-300/-400/070/070</td>
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<td>ROEDER PRAESENTION GMBH</td>
<td>737-300</td>
<td>Installation of Galley G2 Varg p/n 25-30-3154 inside aircraft between STA 332.25 and STA 350.50</td>
<td>- Documentation according to &quot;Vorschriften der Zulassung&quot; Rev. 3 dated 01.01.2007 Compliance Check List BF3- NY-0938-0A Rev. 2 dated 30.01.2007 Declaration of Completion BF3- NY-0938 dated 30.01.07 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>None</td>
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<td>FOKKER SERVICES B.V.</td>
<td>747-8F/-SP/-SR</td>
<td>Introduction of cockpit windows with improved design for Boeing 747/-200/-300/-400/-100 aircraft. GKN p/n 06314 and 06315 as replacement for Boeing p/n 65B27042() and 65B27046() and GKN p/n 06316 and 06317 as replacement for Boeing p/n 65B2704X() and 65B2704Y(). Revision 1: Transfer from GKN to Fokker Services.</td>
<td>- Modification Approval Sheet, ref. C2053, Issue 2, dated 35th April 2016. MPO Supp SAT TV Rev.</td>
<td>None</td>
<td>None</td>
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<td>AIRBUS S.A.S.</td>
<td>A319-115, A-319-133</td>
<td>Airbus Corporate Jet Centre -- Modification CJ20253: A319 SATTV TAILWIND50 ANTENNA Applicability Update A319-115 for installation of antenna structure/insertions and radome for the installation of the Airborne Live TV system TAILWIND 500 previously approved on A319-115</td>
<td>Airbus Corporate Jet Centre -- Modification CJ20253: A319 SATTV TAILWIND50 ANTENNA Applicability Update A319-115 for installation of antenna structure/insertions and radome for the installation of the Airborne Live TV system TAILWIND 500 previously approved on A319-115</td>
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<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>747-400</td>
<td>Reconfiguration of five Boeing 747-400 aircraft to 54J/62W/228Y. Revision 1: Transfer of STC from SR Technics Switzerland LogiNG Engineering Partners GmbH</td>
<td>C172-25-1002-CSC, Revision 0, or later approved revision C172-25-1002-MDL, Revision 0, or later approved revision</td>
<td>The following EASA approved STC's must be installed concurrently within this design change. EASA.A.S.03316; EASA.A.M.A.S.02572</td>
<td>The following EASA approved STC's must be installed concurrently within this design change. EASA.A.S.03316; EASA.A.M.A.S.02572</td>
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<td>Associated Technical Documentation</td>
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<td>10014696</td>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>10014709</td>
<td>REV. 1</td>
<td>2</td>
<td>SA227-BC, SA226-TC, SA227-AC</td>
<td>Installation of Sandel ST3405 TAWS System</td>
<td>Design, Non-significant change: The Certification Basis of the original product and the following additional/alternative warning requirements are applicable to the change. The change has been determined to be non-significant</td>
<td>EASA approved Engineering Order, Doc. No. 2035840-EC, issued, dated 25-Oct-2010, or later approved issue</td>
<td>Not applicable</td>
<td>09/11/2010</td>
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<td>10014699</td>
<td>SABENA TECHNICS BCC</td>
<td>10014709</td>
<td>REV. 3</td>
<td>A330 ALL MODELS</td>
<td>Installation of Aeromobile GSM Connectivity System onboard Airbus A330/400</td>
<td>Installation of non-electrical Floor Path Marking System</td>
<td>Certification, Non-significant change: The Certification Basis of the original product and the following additional/alternative warning requirements are applicable to this certification change</td>
<td>Not applicable</td>
<td>17/09/2010</td>
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<td>AVIATION TRADERS LIMITÉDEUROPE S.A.S.</td>
<td>10014709</td>
<td>REV. 1</td>
<td>747-400 SERIES</td>
<td>Installation of Bendix / King CAS-81 TCAS II System and Dual Bendix TRA-67 Mode S Transponder</td>
<td>The Type and OSD Certification Bases (CB) for the original product and the following additional or alternative airworthiness requirements are applicable to this certification change</td>
<td>Prior to installation of the STC, Boeing SB 747-34-2309 must be reinstated. Prior to installation of this change/repair it must be determined that the installation of this change/repair will introduce no adverse effect upon the aircraft.</td>
<td>09/11/2021</td>
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<td>AVIATION TRADERS LIMITÉDEUROPE S.A.S.</td>
<td>10014709</td>
<td>REV. 1</td>
<td>747-400 SERIES</td>
<td>Installation of RTA-44D VHF Data Radio System</td>
<td>The Type and OSD Certification Bases (CB) for the original product are applicable to this certification change</td>
<td>This STC is approved only for the product configuration as defined in the approved design documentation referred to in the paragraph 4: Specification/Aircraft Modification. Compatibility with other aircraft/engine configurations shall be determined by the installer. Prior to installation of this change/repair it must be determined that the installation of this change/repair will introduce no adverse effect upon the aircraft.</td>
<td>09/11/2021</td>
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<td>AVIATION TRADERS LIMITÉDEUROPE S.A.S.</td>
<td>10014709</td>
<td>REV. 1</td>
<td>747-400 SERIES</td>
<td>Installation of Collins Data Link ACARS System</td>
<td>The Type and OSD Certification Bases (CB) for the original product are applicable to this certification change</td>
<td>Prior to installation of this change/repair it must be determined that the installation of this change/repair will introduce no adverse effect upon the aircraft.</td>
<td>09/11/2021</td>
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<td>10014713</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>10014709</td>
<td>REV. 2</td>
<td>757-200, 757-300</td>
<td>Installation of non-electrical Floor Path Marking System</td>
<td>Installation of non-electrical Floor Path Marking System</td>
<td>This STC is approved only for the product configuration as defined in the approved design documentation referred to in the paragraph 4: Specification/Aircraft Modification. Compatibility with other aircraft/engine configurations shall be determined by the installer. Prior to installation of this change/repair it must be determined that the installation of this change/repair will introduce no adverse effect upon the aircraft.</td>
<td>05/03/2014</td>
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<td>WILTHANSE TECHNIK AG</td>
<td>REV. 2</td>
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<td>ATR 72-212/212A ATR 72-101/102-201-211 ATR 42/200-300/320-400-500</td>
<td>Installation of non-electrical Floor Path Marking System</td>
<td>-Supplemental Type Definition Document STC023-004M, Issue01 Certification Program CP33-06-04M, Issue 01 Compliance Record List CRLL3-06-04M, Issue 01 Airplane Flight Manual Supplement AFMS A-33-06-04M-APM Revision-Instructions for Continued Airworthiness ICA33-06-04M Initial Revision-Eng... ATR42/52 except those aircraft with convertible seats which do not have a fixed lateral position See Airplane Flight Manual Supplement AFMS A-33-06-04M-APM Initial Revision or later approved revisions Instructions for Continued Airworthiness ICA33-06-04M Initial Revision or later approved revisions...</td>
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<td>REV. 1</td>
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<td>757-200</td>
<td>Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval...</td>
<td>Aviation Traders Ltd. ATL Modification ATU-757/152 at Issue 1 dated 10March 2007Aviation Traders Ltd. Master Document List ATU-757/MDO at Issue 1 dated 20 March 2007Airline Traders Ltd. Master Document List ATU-757/MDO at Issue 1 dated 10 March 2007Airline Traders Ltd. Master Document List ATU-757/MDO at Issue 1 dated 10 March 2007</td>
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<td>10014737</td>
<td>AIRLEC AIR ESPACE</td>
<td>REV. 1</td>
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<td>550</td>
<td>Single Pilot Operations and Associated modificationsThis STC is a transfer from Aci Systèmes in Villefranche France toAirlec Air Espace in Langon France.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval...</td>
<td>Aci Systèmes AFM Supplement AC1-00-AFM M-021 Rev 1 Aic Systèmes Compliance Checklist AC1-00-CCLM-021 Rev 3 dated09/05/2007or later revisions of the above listed document(s) approved/accepted under the EASA system. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>FOKKER TECHNIK E.V.</td>
<td>REV. 4</td>
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<td>A319-115</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval The following paragraph (s) at later amendment CSMEL The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval...</td>
<td>AFM Supplement A319CJ 0.01.00 for STC P EASA.A.S. 02937 MMEL Supplement A319-FOM-MMel-S-001 Revision 2, Chapter 6 dated28.09.2017or later revisions of the above listed documents approved by EASA The activation of the Zonal Drier/humidity System is not part of this Major Change The installation of the Zonal Drier/humidity System is certified however the correct implementation and functioning is not certified. Therefore the Zonal Drier/humidity System may not be activated prior to installation.</td>
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<td>DC-9-87 (MD-87) DC-9-83 (MD-83) DC-9-82 (MD-82) DC-9-81 (MD-81)</td>
<td>Installation of flight ID capable XPOR and CTRL</td>
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<td>REV. 1</td>
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<td>Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval...</td>
<td>Master Data List Ref MDL 757/053, issue for later revision of the above listed document(s) approved/accepted under the EASA system. Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>Limitations/Conditions</td>
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<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td>LBA STC TAD015 ISSUE II</td>
<td>A300 B4-605R-622R A300 B4-605I-603/620-622</td>
<td>Installation of Fixed Emergency Locator Transmitter</td>
<td>Certification Basis for the original product and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Summary Document ESGM 238000-001/CO/01/005-02</td>
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<td>FOKKER SERVICES B.V.</td>
<td>REV. 1</td>
<td>737-600/700/800/900  737-100/200-300-400-500-500</td>
<td>Number 2 Cockpit Windshield: Addition of Layers of Polyurethane to the Existing Qualified Design to Improve In-Service Reliability. Rev. 1. Transfer from GKN Aerospace to Fokker Services</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Plan TTN/T/ 07/15 3 General Arrangement Drawing O82405 Issue 7 Report TTN/T/ 8/142/Compliance Checklist CR 016 Issue 02MajorChange Approval CR 016 Issue 02CCM B737 Flight Compartment No. 2 Sliding Window Part No: 08924, 089258 later revisions</td>
<td>None</td>
<td>17/07/2017</td>
<td>Active</td>
<td>10014782</td>
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<tr>
<td>10014784</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY 575-200</td>
<td>Installation of fixed Emergency Locator Transmitter</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Modification BJB866 Technical Publication Supplements BJB866058</td>
<td>N/A</td>
<td>30/07/2019</td>
<td>Active</td>
<td>10014784</td>
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<tr>
<td>10014798</td>
<td>AVIATION ENGINEERING AS</td>
<td>REV. 2</td>
<td>B200</td>
<td>Installation of UNIFIS 3000 Flight Inspection System including antenna beacon B200. The system is used to check and calibrate ground based navigation aids.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Air Alpha Master Drawing No. EN-B200-25-116-01, Rev. 1, dated 30 August 2007 Modification Front sheet for modification EO-B200-25-116 with complete listing of Declaration of Compliance, Report EO-B200-25-116-1 Air Alpha Flight Manual Supplement for Beechcraft B200 with Flight Inspection System UNIFIS</td>
<td>Limited to B200 S/N BB-829. The installation of this modification by third persons is subject to written permission from the approval holder and disposal of the approved/prerequisite documentation. Prior to installation of this modification it must be determined that the relationship between this modification and any other previously installed design change and/or repair will introduce no adverse affect upon the airworthiness of the product</td>
<td>13/03/2019</td>
<td>Active</td>
<td>10014798</td>
<td></td>
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<tr>
<td>10014804</td>
<td>RODER PRAESSION GMBH</td>
<td>REV. 1</td>
<td>425</td>
<td>Installation of GTX 330D Mode S Transponder</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Doc. No. EO C425- NY-1240-05, R5 - EASA approved NAYAK AFM-S Doc. No. C425- NY-1240-05, or later approved revisions - NAYAK Maintenance Procedures. Doc. No. C425- NY-1240-07, or later approved revisions</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>30/10/2017</td>
<td>Active</td>
<td>10014804</td>
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<tr>
<td>10014805</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 1</td>
<td>MYSTERE FALCON 20 ES, 20 GF MFSI, MP900 C550 CITATION II</td>
<td>Cabin Interior Refurbishment applicable to Dassault Aviation Falcon20-ES aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Dossier de Demonstration de Conformite DDC-002-07 Revision 1 or later approved revisions Master Data Unit MOL 050-07 Revision 1 or later approved revisions Aircraft Flight Manual Supplement UAE-AFM-FS-07-01 Original Issue or later approved revisions</td>
<td>This STC is approved only for Mystere Falcon 20-ES aircraft and not for the Mystere Falcon 20-ED aircraft. Prior to installation of this modification/installation must be determined that the relationship between this modification/installation and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product</td>
<td>05/01/2016</td>
<td>Active</td>
<td>10014805</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10014808</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 4</td>
<td></td>
<td>CL-600-2E25 (RU SERIES 1000)</td>
<td>Installation of non-electrical Floor Path Marking - FP-PM Eight FWS: Colours for light strips and carpet protectors, update of documentation, plus addition of model 2E25.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>+ Supplemental Type Definition Document ref. doc. STD03-07/01Rev0, dated 21/3/2017 - Instructions for Continued Airworthiness ref. doc. ICA3-D03-100/1 Rev0, Compliance List ref. doc. CRL3-33-16/00A Rev0, dated 21/11/2017 - Airplane Flight Manual Supplement ref. doc. A-33-07-04</td>
<td>None</td>
<td>21/03/2017</td>
<td>Active</td>
<td>10014808</td>
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<tr>
<td>10014818</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING SERVICES ING. WALTER STARZACHER GMBH</td>
<td>REV. 2</td>
<td></td>
<td>A320 MODELS</td>
<td>Aircraft Installation of an In-Flight Entertainment System (Live TV) on Airbus A320 Aircraft</td>
<td>Certification Basis as per EASA TCDS A.064 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>- Engineering Order, ref. doc. No. EOA_M06800106_CKA_060_82010, Revision 0, dated 4th November 2010 - Instructions for Continued Airworthiness, ref. doc. No.ICA_M06800106_MGR_31052010, Revision 0</td>
<td>None</td>
<td>05/11/2010</td>
<td>Active</td>
<td>10014818</td>
</tr>
<tr>
<td>10014821</td>
<td>JET AVIATION AG, BASEL</td>
<td>REV. 1</td>
<td></td>
<td>A330-202 CSM</td>
<td>A330-202 Cabin Interior Outfitting</td>
<td>Certification Basis as per EASA TCDS A.064 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Plan, ref. doc. No. B3300605, Rev. 0, dated 05.11.2008 - Master Document List, ref. No. B3300605-001, Rev. 0, dated 03.11.2008 - Airplane Flight Manual Supplement, ref. doc. No. B3300605-201-01, Rev. 0, dated 22.10.2008 - Applicant's Declaration of Compliance - Cabin Outfit...</td>
<td>None</td>
<td>02/11/2010</td>
<td>Active</td>
<td>10014821</td>
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<td>10014832</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 3</td>
<td></td>
<td>PA-34-220T (SENECA V)</td>
<td>System PA-34-812000 Installation of Penta Camera System System PA-34-813000 Version 2024425 Camera Hatch and Equipment Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>System PA-34-812000 Flight Manual Supplement SPOH_2024425_rev0_20 11/12/2010 - Master Document List MDOCL_2024425_rev0_20 11/12/2010 - SPOH_14043_rev0_0612/006/2011 - System PA-34-813000 Master Document List Instruction for Continued Airworthiness, ref. doc. No.ICA_M06800106_MGR_31052010, Revision 0</td>
<td>None</td>
<td>03/02/2022</td>
<td>Active</td>
<td>10014832</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10014844</td>
<td>AIR SERVICE VAMDRUP APS</td>
<td>REV. 2</td>
<td>PC-12/47, PC-12/45</td>
<td>Oxygen fill point modification (to remove limitation for installation of STC 10014844 (previous EASA A.S.03062))</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: As per EASA STC 10014844 (former EASA A.S.03062) the requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Engineering Order EO-PC12-31-121, 1 March 2007; - Maintenance Manual EL-PC12-31-121-01: - Aircraft Flight Manual supplement, FMS EO-PC12-31-121-1, Supplemental Maintenance Manual ATP31.014 Rev.0, 21 June 2007; - MEL Supplement EO-PC12-31-121, Report N. MEL0101: When modification &quot;Oxygen fill point...&quot;</td>
<td>03/06/2013</td>
<td>Active</td>
<td>1014844</td>
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<tr>
<td>10014848</td>
<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td>B200</td>
<td>TDR-94D-007 Mode S Transponder Dual Installation.</td>
<td>Installation must be in accordance with AirPlus Master Document List MD-0109-340000 Rev. 2, dated 02.07.07 or later approved. Revisions: Operation must be in accordance with AirPlus Flight Manual Supplement for TDR94D-007, PO-0109-340100 Rev. 1 dated 03.07.07 or later approved. Revision: Maintenance must be in accordance with MD-0109-000000 Rev. 2, dated 02.07.07 or later approved.</td>
<td>None</td>
<td>17/02/2021</td>
<td>Active</td>
<td>1014848</td>
<td></td>
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<tr>
<td>10014849</td>
<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td>T303</td>
<td>Installation of GNSS480 COM/NAV/GPS, TRANSX800, SKY97 Skywatch, KC55A Compass System, SL30 COM/NAV, KN83DME, GWX88 Radar, M350 MFD and GMA 347 Audio System. This installation includes also one Digiflow and Engine monitoring system.</td>
<td>Installation must be in accordance with AirPlus Master Document List MD-0121-340900 Rev. 3 dated 17 Sep 07 or later approved. Revisions: Operation must be in accordance with Flight Manual Supplement PO-0121-340100 to PO-0121-340900 Rev. 1, or later approved. Revision: Maintenance must be in accordance with MD-0121-000000 Rev. 2, dated 02.07.07 or later approved.</td>
<td>None</td>
<td>17/02/2021</td>
<td>Active</td>
<td>1014849</td>
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<tr>
<td>10014857</td>
<td>AIR ALLIANCE GmbH</td>
<td>REV. 4</td>
<td>208, 208B</td>
<td>Conversion of airplane for parachute operations/Conversion of airplane for parachute operations. Installation of PARAKITNAY001 or Parakit C208-83 00 00 &quot;System Gomolzig&quot;</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>For &quot;Parakit NAY001&quot;: According to Master Document List ref: Master Data List PARAKITNAY001&quot;, revision 06, 29.03.1997., To be operated and maintained in accordance with the Applied parachute Flight Manual Supplement ref. &quot;PARAKIT NAY001&quot; revision 06, 29.03.1997 For &quot;Parakit C208-83 00 00 - System...&quot;</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>21/08/2020</td>
<td>Active</td>
<td>1014857</td>
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<td>STC Holder</td>
<td>STC No</td>
<td>Rev. Level</td>
<td>Foreign NAA STC REF</td>
<td>STC Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd</td>
<td>10014883</td>
<td>REV. 4</td>
<td>CASA UK RESPONSIBILITY</td>
<td>B747-400,-747-100,-700C/F,-700</td>
<td>STC EASA.A.S.03087: This modification introduces a Universal Cockpit Display System (Electronic Flight Bag) into the aircraft consisting of 2 Flight Deck Displays, an Application Server Unit, an Accessible Data Unit, an external video camera and associated wiring as detailed in Compliance Document Summa...</td>
<td>Certification basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>21/05/2012</td>
<td>Active</td>
<td>1011662</td>
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<tr>
<td>ROEDER POMEZION GMBH</td>
<td>10014881</td>
<td>REV. 1</td>
<td>MIU-28-40, MIU-28-60</td>
<td>Installation of XDPDR Mode-S System with Diversity, Garmin GTX 330 (single or optional dual installation possible).</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/10/2017</td>
<td>Active</td>
<td>10114881</td>
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<tr>
<td>CESKA LETECKA SERVISNI A.S.</td>
<td>10014884</td>
<td>REV. 1</td>
<td>5600L</td>
<td>Installation of UNIFIS 3000 Flight Inspection System: Console, cockpit Indication system, antenna/This STC is a transfer of EASA STC A.S. 03112 from Air Alpha Maintenance/S.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/06/2017</td>
<td>Active</td>
<td>10114884</td>
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<tr>
<td>AVIATION TRADERS LIMITED/EUROPE S.A.S.</td>
<td>10014887</td>
<td>REV. 1</td>
<td>BAE 146/AVRO 146 MODELS</td>
<td>Installation of UNIFIS 3000 Flight Inspection System: Console, cockpit Indication system, antenna. This STC is a transfer of EASA STC A.S. 03112 from Air Alpha Maintenance/S.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
<td>Active</td>
<td>10114887</td>
<td></td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>10014897</td>
<td>REV. 3</td>
<td>777-200-200LR/1-300-300ER</td>
<td>Installation of non-electrical floor path marking systems/1: System commercial name removed/Rev 3: Extension of existing STC to models 777-200LR and 777-300ER.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Supplemental Type Design Document STO033-0703M Issue 3- AFM Supplement A-33-07/30-1AFM Issue 1- MMEL Supplement A-00-1941-MMEL Issue 00- Instructions for continued Airworthiness ICA325-07/05M Issue/er 01/2012 later revisions of the above listed document(s) approved/accepted under the EASA system...</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/05/2019</td>
<td>Active</td>
<td>10114897</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC ref.</td>
<td>Rev. level</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10014907</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 4</td>
<td>U206, U206A, U206B, U206D, U206E, U206F, U206G</td>
<td>Installation of Cessna 206 Exhaust System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>1, TP206C-E, TU206B, E-G Exhaust System with Extension C1</td>
<td>None</td>
<td>19/01/2022</td>
<td>Active</td>
<td>1014907</td>
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<tr>
<td>10014911</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td>A320-214</td>
<td>Airbus Corporate Jet Centre Modification CJ0103 - VIP Interior &quot;Shining Star&quot; No. 2</td>
<td>Revision due to update of AFM-S and MD-PS</td>
<td>-Modification Approval Sheet ref.: CJ0103 Issue 4, dated 24th October 2013- AFM Supplement: FAM-S MOD CJ0103(&lt;,&gt;); Revision 05, dated 1st May 2012-AMM-S MOD CJ0103(&lt;,&gt;); Revision 02, dated 1st May 2012- AMM-S MOD CJ0103, dated 31st November 2007 or later revisions of the above listed documents approved.</td>
<td>This STC is limited to MSN 09/11/2021. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/11/2013</td>
<td>Active</td>
<td>1014911</td>
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<td>10014924</td>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 2</td>
<td>A330-322</td>
<td>Business Class extension to a 30 B/C + 295 Y/C layout This STC modifies cabin arrangement from a 18 B/C + 343 Y/C layout to a 30 B/C + 295 Y/C layout with new B/C seat and stowage compartment installations.</td>
<td>The Certification Basis for the original product and the following additional or alternative interchangeability requirements as listed in the General CRI G-2102 &quot;ACP Project Certification Basis&quot;: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Master Document List, ref. MD330-25-01-07 Revision 1, dated 25 Oct 2007(&lt;,&gt;) or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2018</td>
<td>Active</td>
<td>1014924</td>
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<td>10014936</td>
<td>SAAB AB</td>
<td>REV. 3</td>
<td>CESSNA 560 560</td>
<td>Replacement of SSR Mode S Transponders for compliance with EnhancedSurveillance Rev. 1: Transfer from Saab Aerotech Aircraft Services to SAABKTEBLOAG Rev. 2: Correction of DOA reference Rev. 3: Correction of Associated Technical Documentation</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>SAAB AB TDM07-18, revision defined in TDM07-18-0-LEP, rev. A, 2007-11-01 AFM Supplement TDM07-18-3-AFM, rev. A 2007-08-15 or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/04/2024</td>
<td>Active</td>
<td>1014936</td>
<td></td>
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<tr>
<td>10014937</td>
<td>SAAB AB</td>
<td>REV. 3</td>
<td>CESSNA 650 650</td>
<td>Replacement of SSR Mode S Transponders for compliance with EnhancedSurveillance Rev. 1: Transfer from Saab Aerotech Aircraft Services to SAABKTEBLOAG Rev. 2: Correction of DOA reference Rev. 3: Correction of Associated Technical Documentation</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Saab AB TDM07-19, revision defined in TDM07-19-0-LEP, Rev. A, 2007-11-01 AFM Supplement TDM07-19-3-AFM, rev. A 2007-08-15 or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/04/2024</td>
<td>Active</td>
<td>1014937</td>
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<td>10014940</td>
<td>AVIATION TRACERS LIMITEDEUROPE S.A.S.</td>
<td>REV. 1</td>
<td>737-735 737-700</td>
<td>148 Pax LOPA Installation Rev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Modification Summary AT0137600 Issue 2- DRN15012-001 Issue 2, or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Compatibility of this design change with other previously approved modifications must be determined by the installer. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/11/2021</td>
<td>Active</td>
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**List of EASA Supplemental Type Certificates**

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<td>10014942</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td>757-200</td>
<td></td>
<td>219 PAX LOFIA Installation Rev. 1: Transfer of STC</td>
<td>The Type and OISD Certification Bases (CB) for the original producetech and the associatedcertification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>AT-Modified Summary Sheet AT/757/180, issue 1, 02/08/2007 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This STC is limited to aircraft w/ n 26270. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
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<td>10014949</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 1</td>
<td>757-200</td>
<td></td>
<td>Introduction of alternative EGPWS Mode 2/3 obstacle Menu Rev 1 from AES Ltd to new STC holder AES Global Limited.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any i...</td>
<td>AFS Modification AES-757-94, issue 2 01/09/2007AES AFM Supplement No. 8 issue 1</td>
<td>FAA STC ST00381BSE must be installed prior to the installation of this STC validated as EASA.IM.A.S.2731 (10014942)</td>
<td>28/04/2021</td>
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<td>10014950</td>
<td>DSA A.S.</td>
<td>REV. 1</td>
<td>T303</td>
<td></td>
<td>COMMNAV/EGPVS GNS 530, GNS 430, Navigation Indicator G10 6, S-mode GTX 330 Transponder, GMA 340 ICs, and ELT Kannad 400AF Installation into Casaga T303</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval</td>
<td>Supplement to Airplane Flight Manual No. 7, 8, 9, 10, 11, 12 ref. 57 LDA02/07/7303 revise 388 LDA 02/07/7303 revise 359 LDA 02/07/7303 revise 351 LDA 02/07/7303 revise 3512 LDA 02/07/7303 revise 3 for later revisions of the above listed document(s) approved/accepted underneath.</td>
<td>S/N: T303 02/07/05. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/06/2020</td>
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<td>10014951</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 1</td>
<td>MYSTERE FALCON 20 ES, 20 GF MF50, MF50 C550 CITATION II</td>
<td></td>
<td>Antenna and Cables Installation provisions for satellite communication system applicable to Dassault Aviation MF50 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval</td>
<td>1. Uni Air Enterprise Master Data List MDL-016-07 Revision 4, dated 10 June 2008.2. Installation instructions in accordance with Uni Air Enterprises Service Bulletin SB-UAE-016-07-01 Revision 3, dated 9 June 2008.3. Uni Air Enterprise Maintenance Manual Supplement UAE-AMMS-07-004Revision 3, dated 2.</td>
<td>Prior to installation of this modification the installer must determine the change/repair and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. T...</td>
<td>05/01/2015</td>
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<td>10014952</td>
<td>airbus engineering GmbH</td>
<td>REV. 1</td>
<td>S550 (CITATION SIII)</td>
<td></td>
<td>Installation of Bendix/King KLN 900, Shadin ADC-200 and Garmin GNS 530A.</td>
<td>• 1. Master Document List, Rev. 01, dated 4 September 2007 or later approved revision. • Maintenance Manual Supplement Navigation System KLN 960, MM-0175-340100, Rev. 0, or later approved revision. • Maintenance Manual Supplement COM/VOR/ILS/GPS/STANS Garmin GNS-530A, MM-0175-340200, Rev. 0 or later.</td>
<td>Prior to installation of this modification the installer must determine the change/repair and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>17/02/2021</td>
<td>Active</td>
<td>1074192</td>
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<tr>
<td>10014953</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 6</td>
<td>PC-12/47PC-12/47PC-12/45PC-12</td>
<td></td>
<td>Installation of MTV-27 (or MTV-47) propeller on Pilatus PC-12 Series equipped with PT6A-67 (or PT6E-67XP engine. Updated AFMS Performance Supplement for Pilatus PC-12 Series equipped w/MTV-47 (or propeller and PT6A-67, or PT6E-67XP engines.</td>
<td>The Type and OISD Certification Bases for the original product are amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-36 Amend 6, FAR 36 Amend 36-1131 this certificate/approval involves a change with an impact on the noise database...</td>
<td>For all PC-12 models with PT6A-67 (or engine and MTV-27 (or propeller According to MT-Propeller Master Document List No. E-1822 Rev. 14 to be operated in accordance with MT-Propeller AFMS Document No. E-1822Rev. 7 to be installed in a w/ MT-Propeller Installation Instruction Doc. No.E-1824 Rev. 10...</td>
<td>According limitation section of AFMS-STP to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/03/2023</td>
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<td>10014968</td>
<td>DSA A.S.</td>
<td>REV. 1</td>
<td>10014967</td>
<td>T208H</td>
<td>Installation of Leica RC30 with IPAS 10 and FCMS options.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
<td>S/N T2068M058 (prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair must be determined and that introduction of such change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/05/2020</td>
<td>Active</td>
<td>10716272</td>
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<td>10014963</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td>10014958</td>
<td>CL-600-2B16</td>
<td>Introduction of Honeywell Mark V EGPWS Rev. 1. Transfer of STC to Magnetic MRO S.A.S.</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
<td>Aviation Traders Ltd Modification ATL/600/003 (EASA.A.S.03251) must be embodied prior to, or concurrently with, Modification ATL/600/002. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
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<td>10014966</td>
<td>MAGNETIC MRO AS</td>
<td>REV. 2</td>
<td>10014967</td>
<td>757-200</td>
<td>Installation of Galexey 1B and Forward RH Wardrobe Revision Transfer of Certificate to Magnetic MRO AS</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
<td>MacCarthy Interiors Modification Leaflet 07ML149.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/09/2019</td>
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<td>10014967</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td>10014971</td>
<td>CL-600-2B16</td>
<td>Installation of Honeywell MK IV EGPWS together with MFD and Thommen Air Data Computer</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
<td>EASA approved „Übersicht der Dokumente“ Project No. SA226/227- NY-0808-Revision 0, dated 07-Nov-2007, or later approved revision/EASA approved Airplane Flight Manual Supplement, Doc. No. SA226/227- NY-0808-08, Revision 0, dated 31-May-2007, or later approved revision/Maintenance Procedures, Doc. No. SA22.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change, repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/11/2021</td>
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<td>10014971</td>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td>10014973</td>
<td>SA227-AT, SA227-TT, SA227-DU, SA226-T, SA226-AT, SA226-DB</td>
<td>Installation of a Dual nav/avionics Electronic and logging display system. Change consisting in installing a dual portable Class 2 EFPL's Bag system.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: ( \text{CRI} ) H-1 dated 31/12/2011; The associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
<td>- Engineering order and certification program 2087424-EO Issue 1 dated 21/05/2011 - AFM supplement 2087424-AM-A Issue 2 dated 14/10/2011; AFM supplement with 3G automatic 2087424-AM-B Issue 2 dated 14/10/2011; Instructions for Continuing Airworthiness 2087424-CA Issue 2 dated 13/10/2011 for later revision...</td>
<td>Not Applicable</td>
<td>14/11/2021</td>
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<tr>
<td>AVIATION ENGINEERING AS</td>
<td>B200</td>
<td>Garmin GXM-68 Radar, Garmin GMX-200 Multi-Function Display, and Avidyne TAWS</td>
<td>The Certification Basis for the original product as amendedthefollowing additional or alternative airworthiness requirements: CS 23 amendment 23-0, CS23.251, 23.301, 23.303, 23.305, 23.307, 23.341, 23.365, 23.561, 23.571a, 23.573b, 23.601, 23.603, 23.605, 23.607, 23.611, 23.613, 23.627, 23.867.</td>
<td>Flight Manual Supplement FMS EO-B200-34-115 Engineering Order (EO-B200-34-118) and related documentation earlier revisions of the above listed documents approved by EASA.</td>
<td>Limited to S/N BB829 The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved documentation. Prior to installation of this modification it must be determined that the interrelationship between this modification...</td>
<td>13/02/2019</td>
<td>Active</td>
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<td>AVIATION ENGINEERING AS</td>
<td>B200</td>
<td>Installation of two Sandel SN3500 EHSI (Electronic Horizontal Situation Indicator) replacing existing HSI displays.</td>
<td>The Certification Basis for the original product as amendedthe following additional or alternative airworthiness requirements: CS 23 amendment 23-0, CS23.301, 23.303, 23.305, 23.307, 23.341, 23.561, 23.601, 23.603, 23.605, 23.607, 23.811, 23.813, 23.1301, 23.1309, 23.1311, 23.1321, 23.1322, 23.1331, 23.1332.</td>
<td>Flight Manual Supplement FMS EO-B200-34-119, Revision 1 and Revision 2 Engineering Order (EO-B200-34-119) and related documents earlier revisions of the above listed documents approved by EASA.</td>
<td>Limited to S/N BB829 The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved documentation. Prior to installation of this modification it must be determined that the interrelationship between this modification...</td>
<td>13/03/2019</td>
<td>Active</td>
<td>10119476</td>
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<tr>
<td>MT-PROPELLER ENTWICKLUNG Gmbh</td>
<td>335, 340, 340A</td>
<td>Installation of 4-blade MT-Propeller Model MTV-14-D-C-FCF168-30g on Cessna 335, 340, 340A.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval: CS 36 Amendment 2 and FAR 36 Amendment 28. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>According to MT-Propeller Master Document List No. E-2460 and MT-Propeller Installation Instructions Doc. No. E-1530, related revision 2 to be operated in accordance with MT-Propeller AFM/AS Document No. E-1520 to be maintained (i.e., MT-Propeller ICA Document No. E-1531) later revisions of the above listed documents.</td>
<td>According to limitation section of AFM. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will increase roadworthiness effect upon the airworthiness of the product. The installation...</td>
<td>19/12/2013</td>
<td>Active</td>
<td>10149693</td>
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<tr>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>757-200</td>
<td>Passenger Cargo Combi Configuration 3: Transfer of STC</td>
<td>The Type and OSD Certification Basis for the original product as amended, the following additional or alternative airworthiness requirements: For changed areas Certification Basis CS 23 amendment 2, including the following: CR D-01, Special Condition: Cargo Compartment Classification CR H-01, Spec.</td>
<td>ATL Master Document List: MOD-11630-00001, Issue 3, 04/02/2011, ATL Modification Summary Sheet: MOD-11630-00001, Issue 2, 04/2020/11, later revisions of the above listed documents/approval/accepted by the EASA system.</td>
<td>This STC is limited to an aircraft with the following aircraft number: 29261 only civil transport aircraft. Modification MOD-11630/00003 must be installed concurrently with this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/or any previously installed change/repair...</td>
<td>09/11/2021</td>
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<td>10149546</td>
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<tr>
<td>SAFRAN SEATS</td>
<td>DC-9</td>
<td>Meridiana DC-9 Stow Bin Extension.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. Requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Health Techia modification record C1037.</td>
<td>Restricted to ASN 49247, 49246, 49531, 49666, 49740, 49911, 49902, 49903, 53049, 49808, 53182, 49785, 49702, 49935, 49951, 49854, 53031. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed changes and/or...</td>
<td>14/03/2019</td>
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<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>FALCON 10</td>
<td>Falcon 10 EHS Mode S Capability EASA A S.03034 has been reassessed with reference 10015004. Rev. 1 following a transfer from ACI Systems to Business &amp; Commuter Aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>AFM Supplement ACI-05-AFM-140/AS-Rev[ACI-05-TD-M-048-Rv0]</td>
<td>Applicability for Falcon 10 that are already Mode A/C Mode C/Mode S and Autels EN 9784 A-07 certified and equipped with a dual transponder system.</td>
<td>02/07/2013</td>
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<td>10015011</td>
<td>LUTTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td></td>
<td>EMB-135/-145 SERIES</td>
<td>Installation of non-electrical Floor Path Marking System</td>
<td>SupPLEMENTAL Type Definition Document STD030-0715M, issue 02 Appliance Flight Manual Supplement A-33-07709- APM Applicant’s Declaration of Compliance dated 20.11.2007 or later revisions of the above listed documents approved by EASA</td>
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<td>10015012</td>
<td>RUAG AG</td>
<td>REV. 1</td>
<td></td>
<td>DHC-6 SERIES 300</td>
<td>Avionics system upgrade: In- Weather Radar System incl. ART-2000, AA-2012, CM-2000, Garmin GMMX-200 Multi Function Display system; Avionix TAS-620 system incl. Heading Adapter AHR6-600, Garmin GTX-330 Mode S Transponder system; Collins DME IND-42 system; UNS-4X FMU Upgrade, Bendix/ King EHSI X-8250M...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any...</td>
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<td>10015022</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 1</td>
<td></td>
<td>CESSNA 425</td>
<td>Alternative Supplemental Inspection Document (A-SID) Rev. 1Transfer to ACC Columbia Jet Service GmbH</td>
<td>- According to Gomolzig MDL Document No. 17070, Revision 04/12 2008 or later EASA approved Revisions.- The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any...</td>
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<td>10015025</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>A320-214</td>
<td>Airbus Corporate Jet Centre Modification CJ0145 - VIP Interior -&quot;Shining Star No. 1&quot;Revision due to update of APM-S andMPD-S</td>
<td>- Modification Approval Sheet ref.: CJ0145 Issue 4, dated 24th October2013- APM Supplement ref.: FM-S MOD CJ0145, Revision 06, dated 14th October 2013- MPD Supplement ref.: FM-S MOD CJ0145, Revision 06, dated 14th October 2013- MPD Supplement ref.: FM-S MOD CJ0277 dated 01/10/08 or later revisions of...</td>
<td>This STC is limited to MSN 3164/Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation...</td>
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<td>10015027</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 5</td>
<td></td>
<td>208, 208 B</td>
<td>Installation of Dual Camera Hatches C208-81200 with sliding door and installation power supply/Installation of Single Camera/Hatch C208-81210 with sliding door and installation power supply</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation...</td>
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<td>10015029</td>
<td>INTERNATIONAL ENGINEERING MASTERS DISTANCE (IEA)</td>
<td>REV. 3</td>
<td>-</td>
<td>PIPER PA-28</td>
<td>Rudder, hand Control and Brake System updated Engineering design drawings.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental/ protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List FAA 37/2007, rev.2 dated 03/06/2008 and as modified by Master Data List: IEA-223 Rev.8, dated 12/12/2018 Flight Manual supplement: AFM-37/2007-02, rev.2 dated 03/06/2008 for Continued Airworthiness ICA-37-2007-000, rev.0 dated 05/10/2007 or later revisions of the above listed d.</td>
<td>Prior to installation of this modification the installer must determine the interrelationship between this modification and any other previously installed modification will introduce no adverse effect on the airworthiness of the product.</td>
<td>12/12/2016</td>
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<td>10015031</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 4</td>
<td>-</td>
<td>DO 28 D-2</td>
<td>Do 28-G92 HT6 Engine Installation/ Turbine Conversion of Do 28 D-2 (model marketing name Do 28-G92) installing variants G92-A-1900-G92-A-1950-G92-B-1800-G92-B-1950-G92-H75-200W with associated implementation of an inspection program to ensure continued airworthiness through early detection of struct...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements or environmental/ protection or a change to the certified noise or emissions levels.</td>
<td>Technical Characteristics and Limitations: The following technical characteristics and limitations are deviant from the basic aircraft apply for variants G92-A-1900, G92-A-1950, G92-B-1800, G92-B-1950, Engine model: Walter M601 D-2 turboprop engines/ Engine Type Certificate: EASA E. 070/ Engine limits...</td>
<td>19/01/2022</td>
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<tr>
<td>10015034</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 2</td>
<td>-</td>
<td>BEECH B99 BEECH 99</td>
<td>Installation of P&amp;W PTBA-34 Engine Flight Manual Supplement SPPH 08869/10202006 rev.0 dated 15/02/2008</td>
<td>This STC is limited to Beech 99 / B99 models equipped with Propeller McCauley 4HP4C4769/94LMA-4 i.a.w. FAA STC SA15351GL.</td>
<td>20/01/2022</td>
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<td>10015036</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 2</td>
<td>-</td>
<td>MYSTERE FALCON 20</td>
<td>Installation of 220 VAC Electrical Power Supply and Distribution in Cabin applicable to Dassault Aviation MF50 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental/ protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Paragraph 21.A109. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/01/2015</td>
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<td>10015037</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 2</td>
<td>-</td>
<td>MYSTERE FALCON 20</td>
<td>Installation of an In-Flight Passenger Entertainment System in Cabin applicable to Dassault Aviation MF50 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental/ protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>STC limited to those Mystere Falcon 50 (MF50) aircraft on which UnioSTC 10015036 Revision 1 “Installation of B20 VAC Electrical Power Supply and Distribution in Cabin” has been embodied. The approval holder shall fulfill the obligations of Part 21, Paragraph 21.A109. Prior to installation of this modification...</td>
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<td>STC no.</td>
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<td>Model(s)</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10015053</td>
<td>STC Holder</td>
<td>REV. 1</td>
<td>Airbus Corporate Jet Centre Modification CJ0222 - VIP Interior - Residence/Reviation due to update of AFM-S and MPD-S</td>
<td>A310-115</td>
<td>The Certification Basis for the original product as amended/repairs: additional or alternative airworthiness requirements as listed in the General CRI A4301-019 &quot;A310-308 Project Certification Basis&quot; CRI H-01: ICA on EWSI. The requirements for environmental protection and the associated certified n...</td>
<td>Modification/Repair Sheet ref.: CJ0222 Issue 3, &quot;Top Modification/Revision&quot;, dated 24th October 2013- AFM Supplement ref.: FM-S MOD CJ0222, Revision 02, dated 1st December 2013- MPD Supplement ref.: MPA-S MOD CJ0222, Revision 04, dated 1st October 2013-MPD Supplement ref.: MPA-S MOD CJ0310 dated 1st M...</td>
<td>This STC is limited to MSN 3356/Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation...</td>
<td>22/11/2013</td>
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<td>10015062</td>
<td>STC Holder</td>
<td>REV. 1</td>
<td>Airbus Corporate Jet Centre Modification C212-300</td>
<td>690A</td>
<td>GPS Antenna Installation.</td>
<td>Installation must be in accordance with AirPlus Master Document List MD-0224-000000 Rev. 1 dated 05 Feb. 06 or later Approved Revisions. Maintenance must be in accordance...</td>
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<td>17/02/2021</td>
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<td>10015090</td>
<td>STC Holder</td>
<td>REV. 1</td>
<td>Iridium Antenna Installation Ancil ST2100.</td>
<td>CL-600-2B16</td>
<td>Installation in accordance with the AirPlus Master Document List MD-0224-000000 Rev. 1 dated 05 Feb. 06 or later Approved Revisions. Maintenance must be in accordance...</td>
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<td>None</td>
<td>30/07/2018</td>
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<td>10015093</td>
<td>STC Holder</td>
<td>REV. 2</td>
<td>MOD 0603: Navigation/Communication (NAV/COM) set replacement and HF Communication set (provisions) Installation This STC replaces a previous NAV/COM system by an up-to-date one, and installs a new HF Communication set (provision). Rev. 1...</td>
<td>C-212-CC (C-212-200)</td>
<td>The requirements for the original product as amended/repairs: additional or alternative airworthiness requirements CS-25 Amendment 4AC.4-13: The requirements for environmental protection and the associated certified noise and emissions levels of the original product were unchanged and...</td>
<td>Master Data List: AE0803MUL Rev. 6 Dated 22. Sept. 2016 For MSN 387 and MSN 1392. HF communication set (subpart of this STC) is not approved (not installed / not operational wiring provisions are accepted) Prior to installation of this design change it must be determined that the interrelationship between this design change and any o...</td>
<td>26/10/2016</td>
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<td>10015096</td>
<td>STC Holder</td>
<td>REV. 1</td>
<td>Installation of SkyEye99 Transmitter Receiver Computer and Directional Antenna NY164 Installation.</td>
<td>B200</td>
<td>Installation in accordance with AirPlus Master Document List Doc-No. MD-0228-000000 Rev. 1, or later approved Revisions. To be operated in accordance with AirPlus Airplane Flight Manual Supplement Doc-No. PO-0228-349990, Rev 0, or later approved revisions. To be maintained i.a.w. AirPlus Flight Manual...</td>
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<td>17/02/2021</td>
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<td>10015097</td>
<td>STC Holder</td>
<td>REV. 2</td>
<td>BlueSky D1000 Satcom installation/The Supplemental Type Certificate EASA.A.S. C3436 installed provisions for further commercial installation BlueSky Network Satcom System D1000 which Intercom Integration option Rev. 1. approval of the activation of the installed device Rev. 2. Transfer of STC.</td>
<td>A310-325 A310-328 A310-322 A310-324 A310-320 A310-338 A310-304 A310-308 A310-318</td>
<td>The Type and OSD Certification Bases for the original product as amended/repairs: the following additional or alternative airworthiness requirements CS-25 Amendment 4AC.4-13: The requirements for environmental protection and the associated certified noise and emissions levels of the product...</td>
<td>Master Drawing List, ref. MDL/A31/017, Issue D4000 Flight Manual Supplement, ref. 11361- AFM-0001, Issue for later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
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<td>EASA STC no.</td>
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<td>Models</td>
<td>Description</td>
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<td>10015103</td>
<td>AIM AVIATION (JECCO) LIMITED</td>
<td>REV. 1</td>
<td>CAAS UK RESPONSIBIL ITY</td>
<td>A330-243</td>
<td>Major Change to STC EASA.A.S.03446 (AIM Major Change MM384-07) made to accommodate decorative finish change to Contour 0995 series Business/Class Seats.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. AIM Special condition for seat heat release requirements (CRSD-432-001) for areas and parts affected by the change. The requirements for...</td>
<td>- AIM MAJOR Change ref. Doc. MM481-10 Issue C dated 21 December 2010. AIM Certification and Compliance Check List, ref. Doc. Appendix to MM481-10 Issue C dated 21 December 2010. Declaration of compliance, page 24 of AIM Certification and Compliance Check List, ref. Doc. Appendix to MM481-10 Issue C dated 21 December 2010.</td>
<td>22/12/2010</td>
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<td>1015103</td>
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<tr>
<td>10015115</td>
<td>TÄBY AIRMAINTENANCE &amp; ACTIEBOLAG</td>
<td>REV. 1</td>
<td>SAAB SF340A SAAB 340B</td>
<td>LEARJET 35A 35A</td>
<td>Rework of Class C eargo role of QC variant. Rev. 1. Transfer of Certificate</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certifications noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any increase in...</td>
<td>S21.FMS-0038 Flight Manual Supplement S21.MME-001 1 Master Minimum Equipment List Supplement S21.TEC-0185 Supplement Manual for Continued Airworthiness S21.TEC-020 6 Weight and Balance Manual later revisions of the above listed document(s) approved and accepted under the EASA system.</td>
<td>17/02/2020</td>
<td>Active</td>
<td>1015115</td>
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<tr>
<td>10015116</td>
<td>SAAB AB</td>
<td>REV. 2</td>
<td>LEARJET 35A 35A</td>
<td>Installation of Honeywell NK VII EGPHS and Mission Inhibit System. Rev. 2. Transfer from SAAB AB - Saab Aerosoch to SAAB AB.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certifications noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>TDM08-06 with revision as described in TDM08-06-01-LEP, Rev. A, 2008-08.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and any other previously installed change/repair and any other previously installed change/repair and any other previously installed change/repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>06/12/2023</td>
<td>Active</td>
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<td>10015122</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD.</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>81/82 Engine Thrust Programming Connector Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Compliance document BAC/27.07/001 Issue 5 dated 22 January 2013 Instructions for Continued Airworthiness S21.TEC-0185 Rev. C, dated 25/01/2008.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved o...</td>
<td>24/01/2013</td>
<td>Active</td>
<td>1015122</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
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<tr>
<td>SAAB AB</td>
<td>2</td>
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<td>Garmin GTX 330/330D (SSR Mode S Transponder) upgrades for compliance with Enhanced Surveillance (EHS) requirements. Rev. 1: Transfer from SAAB Aerotech AB to SAAB and TB6080AD. Rev. 2: Correction of DOA reference.</td>
<td>The Type and OS Certification Bases (CB) for the original products remaining applicable to this certification/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Saab AB TOM03043, with revisions defined in TD0800-03-ALEP, Rev. A, 2008-02-22AF Supplement TD0800-03-3AFMS, Rev A, 2008-01-28.</td>
<td>None</td>
<td>01/12/2023</td>
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<td>HAWKER BEECHCRAFT LST</td>
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<td>Installation of Steep Approach Capability</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft with Honeywell ESPiWS installed installed to Major Change SC0316A at issue 1 or later; - Operated to Airplane Flight Manual Supplement 1 for Flight Manual 140-690007-0005; Aircraft Crew Manual Supplement - Crew Manual - Series 750 - Maintained in accordance with instructions for Continued Airworthiness</td>
<td>None</td>
<td>05/06/2011</td>
<td>Active</td>
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<td>HAWKER BEECHCRAFT LST</td>
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<td>Installation of Steep Approach Capability</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft with Honeywell ESPiWS installed installed to Major Change SC0317A at issue 1 or later; - Operated to Airplane Flight Manual Supplement 1 for Flight Manual 140-690007-0005; Aircraft Crew Manual Supplement - Crew Manual - Series 900XP - Maintained in accordance with Instructions for Continued Airworthiness</td>
<td>None</td>
<td>05/06/2011</td>
<td>Active</td>
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<td>BLUE SKIES AVIATION OY</td>
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<td>Installation of engine Rotax 912 F3 and installation of Propeller Hoffmann HD-V352/170FQ on Cessna Models 150H, F150H, A150K, FA150K, FA150L, FA150M, FA150N, F150L, and F150L only/Installation of engine Rotax 912 S3 (100PS) and installation of Propeller Hoffmann HD-V352/170FQ on all a.m. models...</td>
<td>The Certification Basis for the original product is amended the following additional or alternative airworthiness requirements: The following paragraph(s) is added at a later amendment: FAR 21 Amendment 33. This certificate/approval involves a change to the requirements for environmental protection or a chain...</td>
<td>Aircraft installed with Rotax 912 F3 Cessna 150H, F150H - Flight Manual Operational Instruction Hoffmann No.605 Issue 09.98, USA approved 06.10.1998; Cessna A150K, FA150K, FA150L, FA150M, FA150N, F150L - Flight Manual OperationalInstruction Cessna A150K and FA150K, FA150L and F150L, SDF No. 131, Issue 12/01.1999...</td>
<td>None</td>
<td>18/07/2018</td>
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<td>AERODATA AG</td>
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<td>Installation of Aerodata Flight Inspection System - AD-AFIS-120, Mod. No. B200-11.01, on - AD-AFIS-250, Mod. No. B200-12.01, on - AD-AFIS-250, Mod. No. B200-12.02, on - AD-AFIS-250, Mod. No. B200-14.01</td>
<td>Design non-significant change: The Certification Basis of the original product and the following alternative or alternative airworthiness requirements are applicable to this change: FAR 23 5361 at Amdt. 7 is corrected to comply with the items affected by the change. Env...</td>
<td>For AD-AFIS-120, Mod. No. B200-11.01 EASA approved Master Document List, Installation of Aerodata FIS, Doc No. MDL-A C-126, Rev. 3, dated 27-Nov-2009, or later approved revision EASA approved Airplane Flight Manual Supplement, Doc. No. FA150-09M, Rev. 4, dated 27-Nov-2009, or later approved revision Approv...</td>
<td>For AD-AFIS-120, Mod. No. B200-11.01 Only applicable for aircraft model B200 and B200GT, and when equipped with Collins ProLine 21+/++ and as conditions as defined in Design Description, Installation of Aerodata FIS, Doc. No. DES-AIC-836 Rev. 3, dated 27-Aug-2008, or later approved revision(++)+F...</td>
<td>08/06/2011</td>
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<td>10015140</td>
<td>DIEL AVIATION LAIPHELM GmbH</td>
<td>REV. 2</td>
<td>777-200</td>
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<td>Installation of Galley and Closets on BAW B777-238, Kit P/N 190331-51</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Program and Compliance Check list CP 190331 dated February 27th, 2008 or later revisions of the above listed document(s) approved/accepted.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/07/2019</td>
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<td>10015162</td>
<td>SAAB AB</td>
<td>REV. 1</td>
<td>LEAURET 35A 35A</td>
<td></td>
<td>Mode S Enhanced Surveillance (EHS) Upgrade Rev. 1: Transfer from Saab Aerotech Aircraft Services to SAAB AB.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>AFI Supplement as per TO/MS/02-03-AFM, Issue A (or later approved revisions).</td>
<td>EHS Non-Capable. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/12/2023</td>
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<td>10015194</td>
<td>CESKA LETECKA SERVISNI A.S.</td>
<td>REV. 1</td>
<td>560XL</td>
<td></td>
<td>Printer Stand Installation This is a transfer of EASA A.3.0353 from Air Alpha Maintenance A/S</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-STC Definition Document EO-B001-C560-25-124- EO-134 Supplemental Maintenance Manual EO-134 Drawing List later revisions of the above listed documents approved by EASA A.3.0353.</td>
<td>None</td>
<td>21/09/2017</td>
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<td>10015172</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 1</td>
<td>MYSTERE FALCON 20 ES, 20 GP</td>
<td>C550 CITATION II</td>
<td>Installation of a passenger entertainment audio and video system in the cabin applicable to Dassault Aviation Mystère Falcon 20 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Uni Air Enterprise Master Data List MDL-008-01 Revision 1, dated 15th May 2008. Installation instructions in accordance with Uni Air Enterprise Service Bulletin SB-UAE-008-08-001 Revision 1, dated 16th May 2008. Uni Air Enterprise Airplane Flight Manual Supplement UAE-AMF-08-0222-Revision original dated...</td>
<td>Design change applicable to Mystere Falcon 20 aircraft serial number 311.</td>
<td>05/01/2015</td>
<td>Active</td>
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<tr>
<td>10015184</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 2</td>
<td>ATR 72-212, ATR 72-212 A, ATR 42-500, ATR 42-300, ATR 42-320</td>
<td></td>
<td>Floor To Floor Nets installation on &quot;Quick Change&quot; ATR42 &amp; 72.This STC consists in installing the floor to floor nets instead of the container cargo system. The nets are installed in the container zone defined by ATR through their Class E Quick Change configuration Revision 1 adds some aircraft models...</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment CS-25 at amendment 4paragraphs: 25.23, 25.25, 25.27, 25.29,25.30.1, 25.303, 25.305, 25.561,25.601, 25.603, 25.607, 25.61...</td>
<td>- Master Data List, AEROCONSEIL document No. 4174-01-A-2103-R05, dated 13 October 2020. - Instructions for Continued Airworthiness, AEROCONSEIL document No.4174-01-A-2211-R02,dated 09 October 2020. - Weight and Balance Manual Supplement. AEROCONSEIL document No.4174-01-A-2305-R02dated 13 October 2020.</td>
<td>Limited to those ATR A/C model sharing the following modificationemboded: For ATR 42-300 &amp; ATR 42-320: Optiononoe - ATR Mod 1393 - Refinishing all floor panels to 400 Kg/m², ando - ATR Mod 1769 - Modify actf/s for Cargo Container Loading, ando - ATR Mod 2080 - Modifying Actuating Mechanism fo...</td>
<td>21/10/2020</td>
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<td>10015207</td>
<td>AVONSIS AVIATION ITALIA S.P.A.</td>
<td>REV. 3</td>
<td>500</td>
<td>STC EASA A.5.0630 dated 2 June 2009 &quot;Satellite Telephone System&quot;Rev. 2 is an administrative re-issuance to reflect the recent name change from INASAero Aviation Italia S.p.A. Divisione Ala Pista. Rev. 3: Transfer of STC</td>
<td>The Type and OSD Certification Bases for the original product as amended the following additional or alternative airworthiness requirements:CS-25. 1309The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain app...</td>
<td>Master Drawing List MDL-23-00-0802-A dated 26 May 2009</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>650 (CITATION III)</td>
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<td>10015219</td>
<td>airplus engineering GmbH</td>
<td>PA-42 (CHEYENNE II)</td>
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<td>10015221</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td>Primus 880 WX &amp; MFMD Installation/Removed existing Weather Radar Equipment - i.a.w. Installation Instructions, Doc. No. OPM-004Rev.- Master Document List, Doc. No. E-2085, or later revisions of the above listed documents approved by EASA.</td>
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<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td>MK VII EGPWS Installation/Removed existing Equipment i.a.w. Installation Instructions, Doc. No. OPM-004New</td>
<td>10015222</td>
<td>23/02/2012</td>
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<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td>RAAS activation on the Honeywell MK-VII EGPWS a.w. Installation Instructions, Doc. No. OPM-005Revision 1 has been issued to reflect the transfer of STC.</td>
<td>10015223</td>
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<td>10015224</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td>P-RNAV Compliance in installed GNS-XL(8) FMS, i.a.w. Installation Instructions, Doc. No. OPM-006Revision 2 has been issued to reflect the removal of the limitation in OPM-006 Citation Service (GNS-XL(8)) FMS, i.a.w.</td>
<td>10015224</td>
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### List of EASA Supplemental Type Certificates

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<th>Rev. level</th>
<th>Model</th>
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<td>10015227</td>
<td>BUCHER LEICHTBAU AG</td>
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<td>REV. 5</td>
<td>PC-12/47E</td>
<td>Bucher EMS Equipment PC-12/47E - P/N 0127001-400</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Compliance with Vendor Service Bulletin 0127001-254-00 Prior to installation of this modification the installer must determine the interrelationship between this modification and any other previously installed modification which will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/06/2015</td>
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<td>10015236</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>-</td>
<td>REV. 1</td>
<td>757-200</td>
<td>Introduction of Flight Control Computer (FCC) Interlock Rev.1</td>
<td>The installation of this modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this modification the installer must determine the interrelationship between this modification and any other previously installed modification which will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10015240</td>
<td>airbus engineering GmbH</td>
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<td>REV. 1</td>
<td>B200</td>
<td>Installation of Dual Garmin COM/NAV/GPS GNS430A/W or GNS430Wand or Avionix EX500 MFD or interfacing with existing WX Radios and/or L-3 Com A1005 Cockpit Voice Recorder and/or Single Counter Pointer Drum Altimeter Aeronetics P/N 10456-1195 and/or and/or and/or...</td>
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<td>10015241</td>
<td>BUSINESS AND COMPUTER AIRCRAFT</td>
<td>-</td>
<td>REV. 2</td>
<td>MYSTERE FALCON 20</td>
<td>Indium telecommunication system installation applicable to Cessna/Embraer aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Installation limited to those Mystere Falcon 50 (MF50) aircraft previously modified i.a.w. Cessna/Embraer aircraft</td>
<td>05/01/2015</td>
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<td>10015249</td>
<td>ALAMO ENGINEERING GmbH</td>
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<td>REV. 3</td>
<td>JETSTREAM 4100 SERIES</td>
<td>Mode S Enhanced Surveillance (EHS) Upgrade Rev. 3</td>
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<td>Prior to installation of this modification the installer must determine the interrelationship between this modification and any other previously installed modification which will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>AVIATION TRADERS LIMITED</td>
<td>PA-34-200 /-200T/-220T</td>
<td>Installation of a Solid State Cockpit Voice Recorder and Control Unit in Boeing 747-100 and -220 series aircraft per ATLMaster Document ListMDL-11053-0001f at issue 1 or later approved revisions Rev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product subschematics applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List Ref MDL-11053-0001f or later approved document(s) approved/accept/replace the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined whether the modification/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third parties...</td>
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<td>LEONE EUROPE</td>
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<td>JET AVIATION AG</td>
<td>GV</td>
<td>Iridium Antenna Installation at dual Iridium antenna has been installed on top of the fuselage located between Sta. 535 &amp; Sta. 554.5.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>Prior to installation of this modification the installer must determine whether the modification/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by...</td>
<td>15/06/2016</td>
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<tr>
<td>ACC COLUMBIA</td>
<td>PA-34-200 /-200T/-220T</td>
<td>Replacement of existing Garmin COM/NAV and existing Garmin GPS system by Garmin GNS 530x COM/NAV/GPS system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Approved Engineering Order, Doc. No. PA34- NY-10118-05, Revision 1 dated 26 January 2009, or later approved revision. - EASA approved Airplane Flight Manual Supplement, Doc. No. PA34- NY-10118-08, Revision 2 dated 22 December 2008, or later approved revision. - Maintenance Procedures, 0.</td>
<td>1. Limited to “Single GNS” Installation only. Prior to installation of this design change it must be determined whether the modification/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by...</td>
<td>30/10/2017</td>
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### List of EASA Supplemental Type Certificates

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<th>STC No.</th>
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<th>Description</th>
<th>Certification basis</th>
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<tr>
<td>10015295</td>
<td>STC TWENTY ONE LIMITED</td>
<td>(EASA.A.S.03727)</td>
<td>BOEING 747-400 SERIES 747-400</td>
<td>Installation of 3 AD Aeroospace FV 0315 External Cameras and 1 AD Aeroospace FV 6018 Video Transmission UnitRev. 1 transferred from Tenencia Ltd. to STC Twenty One Ltd.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. There are no environmental protection and the associated certified noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>14/11/2022</td>
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<td>10015290</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CO.</td>
<td>(EASA.A.S.03806)</td>
<td>206 &amp; 210 (CENTURION)</td>
<td>Installation of Aspen Avionics EFD 1000 system (STC is cover row Cesarac-206 &amp; newly added Cessna C-210 series)</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. Minor Change to the associated certified noise and/or emissions levels of the product is unchanged and remain applicable to this certificate/approval.</td>
<td>26/02/2010</td>
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<tr>
<td>10015281</td>
<td>AIRBUS S.A.S.</td>
<td>(EASA.A.S.03813)</td>
<td>A300C4-203</td>
<td>- Ref. to former Luftfahrt Bundesamt STC 2830/0009 Issue 1 - This EASA STC keeps the certified design of the LBA STC 2830/0009 Issue 1, which consisted in the installation of structure and system provisions to change the loading system to an integrated side by side / single row system, and reflects...</td>
<td>- The Certification Basis for the original product remains applicable to this certificate/approval. There are no environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>26/02/2010</td>
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<td>AIRBUS S.A.S.</td>
<td>(EASA.A.S.03814)</td>
<td>A300F4-605R</td>
<td>- Ref. to former Luftfahrt Bundesamt STC TA0466- This EASA STC keeps the certified design of the LBA STC TA0466, which consisted in the installation of the Lower Deck Cargo Compartment (L SCC) - Cargo Loading System (CLS), and reflects the changes induced by the transfer of LBA STC TA0466 from Airbus Deuts...</td>
<td>- The Certification Basis for the original product remains applicable to this certificate/approval. There are no environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>26/02/2010</td>
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<td>AIRBUS S.A.S.</td>
<td>(EASA.A.S.03816)</td>
<td>A319-132</td>
<td>Equipment - Furnishing - Passenger Compartment - MP 40 for A319/320 Group Keesaar Revision to Luftfahrt Bundesamt STC TA0217 Issue 2. This EASA STC keeps the certified design of the LBA STC TA0217 Issue 2 which consisted in the aircraft cabin reconfiguration, and reflects the changes induced...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. See Certification Documentation for Associated Technical Documentation section. The certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.</td>
<td>12/06/2010</td>
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<td>AIRBUS S.A.S.</td>
<td>(EASA.A.S.03817)</td>
<td>A319-133</td>
<td>Equipment - Furnishing - Passenger Compartment - Install cabin finishing for Damier Chrysler Aviation/Revision to Luftfahrt Bundesamt STC TA0267This EASA STC keeps the certified design of the LBA STC TA0231 which consisted in the aircraft cabin reconfiguration, and reflects the changes induced by...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. See Certification Documentation for Associated Technical Documentation section. The certified noise and/or emissions levels of the product...</td>
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<td>AIRBUS S.A.S.</td>
<td>LBA STC NO. TA0269</td>
<td>Equipment / Furnishing - Passenger Compartment - Install baggagecontainer for Gurteneckpulser Aviation</td>
<td>The Certification Basis for the original product and the following: Aircraft noise, maintenance, repair and storage.</td>
<td>DASA Engineering Order Nr. CBA00100A, Issue 2 or later revisions of the above listed documents approved by EASA</td>
<td>Limited to MSN 1053</td>
<td>12/05/2010</td>
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<td>LBA STC NO. TA0231</td>
<td>Equipment/ Furnishings - Passenger Compartment - Vip kit in the aircraft cabin</td>
<td>The Certification Basis for the original product and the following: Aircraft noise, maintenance, repair and storage.</td>
<td>DASA Engineering Order Nr. SYR037/98A or later revisions of the above listed documents approved by EASA</td>
<td>Limited to MSN 866, 918, 1032, and 1076</td>
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<td>AIRBUS S.A.S.</td>
<td>LBA STC NO. TA0228</td>
<td>Equipment/ Furnishings - Passenger Compartment - Vip kit in the aircraft cabin</td>
<td>The Certification Basis for the original product and the following: Aircraft noise, maintenance, repair and storage.</td>
<td>DASA Engineering Order Nr. SYR010/98A or later revisions of the above listed documents approved by EASA</td>
<td>Limited to MSN 240, 249, 253, 255, 262, 275, 288, 291, and 366</td>
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<td>AIRBUS S.A.S.</td>
<td>LBA STC NO. TA0261</td>
<td>GLARE demonstrator panel installation :Daimler Chrysler Aerospace to Airbus SAS</td>
<td>Certification Program : EASA/44/630/01 - Glare panel installation :Daimler Chrysler Aerospace to Airbus Engineering Order Nr. GAF06698A, Issue 02- Glare panel replacement :Airbus Engineering Order Nr. GAF030/08A, Issue 01 or later revisions of the above listed documents approved by EASA</td>
<td>Certification Program : EASA/44/630/01 - Glare panel installation :Daimler Chrysler Aerospace to Airbus Engineering Order Nr. GAF06698A, Issue 02- Glare panel replacement :Airbus Engineering Order Nr. GAF030/08A, Issue 01 or later revisions of the above listed documents approved by EASA</td>
<td>Limited to MSN 484.</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td>REV. 1</td>
<td>Installation of 149 Pax Layout of Passenger Accommodation (LOPA)</td>
<td>Certification Program : EASA/44/630/01 - Glare panel installation :Daimler Chrysler Aerospace to Airbus Engineering Order Nr. GAF06698A, Issue 02- Glare panel replacement :Airbus Engineering Order Nr. GAF030/08A, Issue 01 or later revisions of the above listed documents approved by EASA</td>
<td>Certification Program : EASA/44/630/01 - Glare panel installation :Daimler Chrysler Aerospace to Airbus Engineering Order Nr. GAF06698A, Issue 02- Glare panel replacement :Airbus Engineering Order Nr. GAF030/08A, Issue 01 or later revisions of the above listed documents approved by EASA</td>
<td>Limited to MSN 484.</td>
<td>12/05/2010</td>
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<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 1</td>
<td>BCA112 revision 2, dated 15 July 2012</td>
<td>The Certification Basis for the original product remains applicable to this certificate / approval.</td>
<td>Certification Program : EASA/44/630/01 - Glare panel installation :Daimler Chrysler Aerospace to Airbus Engineering Order Nr. GAF06698A, Issue 02- Glare panel replacement :Airbus Engineering Order Nr. GAF030/08A, Issue 01 or later revisions of the above listed documents approved by EASA</td>
<td>Certification Program : EASA/44/630/01 - Glare panel installation :Daimler Chrysler Aerospace to Airbus Engineering Order Nr. GAF06698A, Issue 02- Glare panel replacement :Airbus Engineering Order Nr. GAF030/08A, Issue 01 or later revisions of the above listed documents approved by EASA</td>
<td>Limited to an BB-1473</td>
<td>25/07/2012</td>
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**Certification Basis:**
- Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.
- Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.

**Limitations/Conditions:**
- Not applicable
- No applicable
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<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
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<td>10015315</td>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA ST02064AT</td>
<td>727-100, 727-200</td>
<td>Installation of a Honeywell (formerly Aikeyt Signal) 18 Channel Flight Data Recorder (Shingles) for Aircraft not containing a Flight Data Acquisition Unit. Revision 2 transferred from EMTEQ Engineering to B/E Aerospace. The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA ST02064AT</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>02/03/2016</td>
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<td>10015316</td>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA ST03964AT</td>
<td>767-300</td>
<td>Installation of a passenger cabin interior arrangement (2907), Revision 1 transferred from EMTEQ Engineering to B/E Aerospace. The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA ST03964AT</td>
<td></td>
<td>Limited to 767-33A, Serial No. 25535. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>02/03/2018</td>
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<td>10015318</td>
<td>GLOBAL AIRWORKS, Inc.</td>
<td>REV. 2</td>
<td>FAA STC ST01626LA</td>
<td>CL-600-2E25 (RU SERIES 1000)</td>
<td>Installation of AirWorks Cockpit Door Surveillance System. The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA STC ST01626LA</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>05/02/2016</td>
<td>Active</td>
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<td>10015321</td>
<td>S-TEC CORPORATION</td>
<td>REV. 2</td>
<td>FAA STC SA09656W-O</td>
<td>PA-28-236</td>
<td>Installation of S-TEC System 55/55 Two Axes Automatic Flight Guidance System Model ST-666. The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>FAA STC SA09656W-O</td>
<td></td>
<td>None</td>
<td>14/09/2010</td>
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<td>10015322</td>
<td>AVIATION PARTNERS BOEING WINGLET S COMPANY, LLC</td>
<td>REV. 18</td>
<td>FAA STC ST00650E</td>
<td>737-700-800-900-900ER</td>
<td>Installation of Aviation Partners Boeing winglets (Blended Winglets design-Split Sclimbar Winglets) Revision 1 approved 137-950 performance data when CFM56-7B27 engines are installed; Revision 2 approved 737-800 performance data when CFM56-7B27 engines with bump rating are installed; Revision 3, as per...</td>
<td>FAA STC ST00650E</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>04/10/2019</td>
<td>Active</td>
<td>10015322</td>
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- Blended Winglets: The baseline of the certification basis for these approvals is as follows: JAR 25, change 14, effective 27th May 1994; Orange Paper 961, effective 19 April 1996; JAR 70, change 2, effective 1 August 1996; JAA IL-23, effective 21 April 1994; 190473-700: "1", "2", "3", "4", "5", "6", "7", "8". Additional...
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<th>Limitations/Conditions</th>
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**Note:** This page is a partial list of EASA Supplemental Type Certificates. The full list can be found at [easa.europa.eu](https://easa.europa.eu).
<table>
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<th>STC Holder</th>
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<tr>
<td>AIRWORKS, Inc.</td>
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<td>ATR 72-201, ATR 72-202, ATR 72-101, ATR 72-102, ATR 42-320, ATR 42-320</td>
<td>Installation of Hawkeye Mark V Enhanced Ground Proximity Warning System (EGPWS)</td>
<td>Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>757-200</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/03/2016</td>
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<td>10015406</td>
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<td>AIRWORKS, Inc.</td>
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<td>A321-231-232, A321-111-112, A320-211-120, A320-211-212, A320-212-213, A320-213-214</td>
<td>Installation of a Kannad Model 406-AF Emergency Locator Transmitter (ELT)</td>
<td>Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>757-200</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/03/2016</td>
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<tr>
<td>AIRWORKS, Inc.</td>
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<td>777-200, 777-300, 777-3000</td>
<td>Installation of AirWorks Cockpit Door Surveillance System</td>
<td>The Certification Basis (CB) for the original product remains applicable with certification approval.</td>
<td>757-200</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/03/2016</td>
<td>Active</td>
<td>10015416</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
<th>Rev.</th>
<th>Model</th>
<th>Certification Basis</th>
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<tr>
<td><strong>AVIONICS GROUP INC.</strong></td>
<td>REV. 1</td>
<td>FAA ST02186AT</td>
<td>757-200, 757-200PF</td>
<td>Installation of a Tekdyne Digital Flight Data Acquisition Unit, PartNumber 2233000-611</td>
<td>The Certification Basis for the original product and the following(additional or alternative) airworthiness requirements are applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels...</td>
<td>08/02/2011</td>
<td>Active</td>
<td>10015426</td>
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<td><strong>STG AEROSPACE Inc.AKA SAFTGLO, Inc.</strong></td>
<td>REV. 1</td>
<td>FAA STC ST0525AT</td>
<td>ATR 72-212/212A ATR 72-101/201-202/21/21 1 ATR 42-200/300/320-500</td>
<td>Installation of SaftGlo Photoluminescent Floor Proximity Emergency/Escape Path Marking System (PFPEEMPS). Validation of FAA STC ST0525AT (The STC installs a photoluminescent aisle path marking system(PFPEEMPS). This revision of a) ATR T2 aircraft models in the STC applicability.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>08/02/2011</td>
<td>Active</td>
<td>10015430</td>
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<tr>
<td><strong>STG AEROSPACE Inc.AKA SAFTGLO, Inc.</strong></td>
<td>REV. 1</td>
<td>FAA STC ST0406AT</td>
<td>F28 MARK 4000 F28 MARK 3000 F28 MARK 2000 F28 MARK 1000 F28 MARK 0100</td>
<td>Installation of SAF-T-GLO photoluminescent floor proximity emergency/escape path marking system (PFPEEMPS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>03/11/2014</td>
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<td><strong>ROCKWELL COLLINS, Inc.</strong></td>
<td>REV. 3</td>
<td>FAA STC ST00673W-D</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Modification of VHF-23A/VHF-22B VHF COMM/HF COMM Receiver Transmitter with CTL-22 Control Panel.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change must be determined by the installer. If other aircraft/engine paragraph defined in the approved design change, and any interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the original product.</td>
<td>05/01/2017</td>
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<td><strong>PRECISION CONVERSIONS, LLC</strong></td>
<td>REV. 1</td>
<td>FAA STC ST1529SE</td>
<td>757-200</td>
<td>Conversion of a Boeing 757-200 series passenger configuration to airlighter configuration (including main deck cargo door, crew entry/door, interior, and associated systems).</td>
<td>The Certification Basis for the original product as amended by following additional or alternative airworthiness requirements: Equivalent Safety Filing(s): CRI D-02 Carryage of &quot;Supernumeraries&quot; in Flight Compartments JAR 25 Change 15 with the following paragraphs at later amendments CS 25 Amendment 1...</td>
<td>30/04/2017</td>
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<td><strong>ISRAEL AEROSPACE INDUSTRIES LTD.IAI</strong></td>
<td>REV. 5</td>
<td>B737-300</td>
<td>Conversion from Passenger configuration to Special Freighter (SF),Special Freighter with Flight Barrier and to Quick Change (QC)configuration and approval of increased weight limitations. Summary of EASA STC Revisions Original EASA approval: CA-AUK STC No CAA 21NE1.00923T EASA IM.A.S.01426 Revision 1...</td>
<td>EASA STC Certification Basis, as recorded in EASA CRI A- OI dated 2007-11-02 The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Design Document TR 385-210100-26100 Revision &quot;BB&quot;, dated October 28, 2010. - List of added/converted airworthiness TR 385-00-00-03922 Revision &quot;AE&quot;, dated September 18, 2007 or later revisions of the above listed documents approved by EASA.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft configurations shall be predicated by the installer. If the holder agrees to permit another person to use this certificate...</td>
<td>23/04/2012</td>
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<td>Associated Technical Documentation</td>
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<td>LIFEPORt, LLC</td>
<td>(EASA.IM.A.S.01445)</td>
<td>REV. 2</td>
<td>FAA STC SA00273WI</td>
<td>SEE FAA AML SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Stretcher, Patient Loading, and Support system. Rev. 2 adds the Taxi control mode to the subject STC/approved model list.</td>
<td>Lifepor, Inc. Master Drawing List 520-0000, Revision C, dated June 27, 1994, or later FAA Approved revision Airplane Manual Supplement (AFMS) 35 and Supplement No. 5 with all FAA Approval date of December 28, 1994, or later FAA approval revision.</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/02/2022</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>(EASA.IM.A.S.01445)</td>
<td>REV. 2</td>
<td>FAA STC SA06929AC</td>
<td>690D, 695, 695A, 690C</td>
<td>Installation of the S-TEC 01304 Digital Flight Control System (DFCS); in accordance with Master Data List 921128, rev. 12.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>S-TEC Master Data List 921128, Rev. 12, dated May 21st, 2014; Airplane Flight Manual Supplement No. 922134, Rev. 3, dated May 21st, 2014; later revisions of the above listed documents approved by EASA inconsistency with EASA ED Decision 2004/04/C1F (or subsequent revisions of this decision) and/or...</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>FLINT AERO, Inc.</td>
<td>(EASA.IM.A.S.01457)</td>
<td>REV. 1</td>
<td>FAA STC SA2323NM</td>
<td>T206F, T206G, U206F, U206G, U206H</td>
<td>Installation of 14.9 gallon usable fuel contained in lightweight-resistant tip tanks revision 1 adds the model T206F.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection are unchanged but this certificate/approval involves a change with an impact on the notepad database.</td>
<td>Installation in accordance with FAA sealed Flint Aero, Inc., Master Drawing List FAA5206, no revision, dated March 4, 1986. For operation at increase in the takeoff gross weight (Swing/Combi) from 3,600 to 3,800 pounds for amphibian, skiplane, and amphibian versions of noted models. That...</td>
<td>Any FAA STC's referenced below that may be used in conjunction with this STC, are not automatically accepted by EASA it is the Aircraft Operator's responsibility upon the airworthiness of the product.</td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>(EASA.IM.A.S.01470)</td>
<td>REV. 1</td>
<td>FAA STC SA00202AT</td>
<td>BAE JETSTREAM 3201 BAE JETSTREAM 3101</td>
<td>Installation of Sandel Model ST3400 Terrain Awareness and Warning System (TAWS) Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. EMTEQ Engineering Approved Model List (AML) Document No. 2-25975-002 Rev. (R) dated January 14, 2005 or later approved revision. 2. EMTEQ Engineering AFM Supplement Document No. 2-25975-009 Rev. (R) dated January 14, 2005 or later approved revision.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>(EASA.IM.A.S.01475)</td>
<td>REV. 1</td>
<td>FAA STC ST0293AT</td>
<td>BEECH 1900D</td>
<td>Installation of Honeywell MK VI Enhanced Ground Proximity Warning System (EGPWS) Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. EMTEQ Engineering Master Drawing List EMTD-010-MDL, Revision C, dated 07 July 2004 or later FAA approved revision. 2. FAA Approved Flight Manual Supplement for Raytheon Aircraft Company Model 1900D Aircraft with Honeywell Model MK VI EGPWS, EMTEQ Document No. 025-028-001 Rev. (R), dated February...</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>B/E AEROSPACE, Inc.</td>
<td>(EASA.IM.A.S.01476)</td>
<td>REV. 1</td>
<td>FAA STC ST0302AT</td>
<td>EMB-120</td>
<td>Installation of Sandel Model ST3400 Terrain Awareness and Warning System (TAWS) Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. EMTEQ Engineering Master Drawing List (MDL) 2-25975-001 Rev. (R) dated January 6, 2005; 2 FAA Approved Flight Manual Supplement (AFMS) for Embraer EMB-120 Series Aircraft, EMTEQ Document No. 2-25975-001 Rev. (R), dated January 20, 2006.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10015670</td>
<td>(EASA.IM.A.S.01477)</td>
<td>REV. 1</td>
<td>FAA STC 92612-A</td>
<td>BEECH 1900D</td>
<td>Installation of Stretched Model ST3400 Terrain Awareness and Warning System (TAWIS), with optional capability as a Traffic Collision Avoidance System (TCAS) Control Display Unit (CDU) or Radio Magnetic Indicator (RMI); Revision 1 transferred from EMTEQ to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. NEW-PAGE The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. EMTEQ Engineering Master Drawing List EMTEQ-036-MDL, Revision B, dated March 19, 2014 or later FAA approved revision. 2. FAA Approved Flight Manual Supplement (AFMS) for Raytheon AircraftCompany Model 1900 Series Aircraft installed as a TAWS/TCAS CDU. EMTEQ Document No. EMTEQ-09A-0MS, Revision a, dated March 19, 2014.</td>
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<td>B/E AEROSPACE, Inc.</td>
<td>10015571</td>
<td>(EASA.IM.A.S.01478)</td>
<td>REV. 1</td>
<td>FAA STC 700303-A</td>
<td>767-200, 767-300</td>
<td>Installation of a Honeywell Mark V Enhanced Ground Proximity Warning System (EGPWS) Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplements (AFMS) for Boeing 767 series aircraft. EMTEQ Document No. 2-26425-906, Rev. JR, dated December 10, 2004 or later revisions of the above listed documents approved by EASA. EMTEQ Engineering Master Drawing List (MDL) 2-26425-201, Rev. A, dated December 05, 2004.</td>
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<td>AVIATION FABRICATORS, Inc.</td>
<td>10015576</td>
<td>(EASA.IM.A.S.01487)</td>
<td>REV. 1</td>
<td>FAA STC 82602-C</td>
<td>2. 300, 300LW, B200, B200GT 2. 200, 200C, B200CT, B200GT 1. B300, B300C 1. B200, B200GT, B200CT</td>
<td>Installation of two place side-facing divan.</td>
<td>FAR 23</td>
<td>Aviation Fabricators Drawing Number D10002B approved on 30 May 1990 or subsequent FAA approved revision.</td>
</tr>
<tr>
<td>AVIATION FABRICATORS, Inc.</td>
<td>10015577</td>
<td>(EASA.IM.A.S.01488)</td>
<td>REV. 1</td>
<td>FAA STC 80622-W</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Installation of shoulder harness kit</td>
<td>CAR 3 and FAR 23</td>
<td>FAA approved Drawing List AF-203 F, dated 6 October 2009 or later approved revision.</td>
</tr>
<tr>
<td>AIM AVIATION RENTON, INC.</td>
<td>10015600</td>
<td>(EASA.IM.A.S.01536)</td>
<td>REV. 1</td>
<td>FAA STC 14906-E</td>
<td>757-200 SERIES</td>
<td>Installation of a flight deck lavatory.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1) AIM Aviation Inc. Master Data List AIM-14485 Revision A, dated 17 December 2004, or later FAA approved revision. 2) AIM Aviation, Inc. Drawing List AIM-14485, Revision A, dated 17 December 2004, or later FAA approved revision.</td>
</tr>
<tr>
<td>AAR CARGO SYSTEMS</td>
<td>10015627</td>
<td>(EASA.IM.A.S.01579)</td>
<td>REV. 1</td>
<td>FAA STC ST02012CH</td>
<td>MD-11</td>
<td>Installation of a Main Deck Cargo Handling System (FAA STC ST02012CH)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>AAR Report 1826 Instruction for Continued Airworthiness</td>
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<tr>
<td>LB AIRCRAFT ENGINEERING LLC</td>
<td>10015635</td>
<td>(EASA.IM.A.S.01591)</td>
<td>REV. 2</td>
<td>FAA STC 09589AT</td>
<td>737-300-400-500</td>
<td>Installation of Collins Model 700 High Frequency (HF) Communications System and HF antennasRevision 1: Transfer of STC 10015635 (IMA.01591) from ARC Avionics Corporation to ARCAIRCRAFT RADIO &amp; AVIONICS, LLCRevision 2: STC Transfer from Aircraft Radio &amp; Avionics, LLC to &quot;LBAircraft Engineering, LLC&quot;.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>-ARC Avionics Master Drawing List, Ref. ARC-HF-001-MDL, Rev. E, dated 18 December 2005 -ARC Avionics Airplane Flight Manual Supplement, Ref. ARCHF-001-MFS, dated 28 May 2006 or later revisions of the above listed documents approved by EASA in accordance with EASA EC Decision 2004/04/CF (or subsequent revisions).</td>
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<td>STC Holder</td>
<td>STC</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical</td>
<td>Limitations/Conditions</td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>FAA STC 02098AT</td>
<td>REV. 2</td>
<td>FAA</td>
<td>MD-88/90, MD-11 SERIES, DC-10-30 SERIES, 747-100-200 SERIES, 757-200 SERIES, 767-200 SERIES, 777-200 SERIES</td>
<td>Installation of an FAR Part 145 EASA.125 SERIES 700A/B Aircraft Completion, 13 Passenger</td>
<td>FAA Approved Mod Kit (AML,No. 2-22566-002 Rev. K, 12/08/2011)</td>
<td>FAA Approved Documentation (AML,No. 2-22566-002 Rev. K, 12/08/2011)</td>
<td>The STC is valid for those aircraft types and models approved by EASA/ FAA Approved Model List (AML,No. 2-22566-002, Rev. K, 12/08/2011) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed de...</td>
</tr>
<tr>
<td>BOMBARDIER INC.</td>
<td>FAA STC 02011AT</td>
<td>REV. 1</td>
<td>TCCA STC C-LSA05-391-10</td>
<td>BD-700-1A10</td>
<td>Aircraft Completion, 13 Passenger</td>
<td>EASA.125 SERIES 700A/B Interior Configuration</td>
<td>Installation of a Rockwell Collins</td>
<td>FAA Approved Mod Kit (AML,No. 2-22566-002 Rev. K, 12/08/2011)</td>
</tr>
<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>FAA STC 02019AT</td>
<td>REV. 1</td>
<td>FAA</td>
<td>767-300</td>
<td>Installation of Passenger Cabin Interior Arrangement Revision 2 transferred from EASA.125 SERIES 700A/B Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Avions Engineering Master Drawing List (MLD) No. 99GEC3.015MDL Revision E, 17th February 2000.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aerodynamics of the product.</td>
</tr>
<tr>
<td>328 SUPPORT SERVICES GmbH</td>
<td>HS.125 SERIES 700A/B</td>
<td>REV. 3</td>
<td>HFS 125 SERIES 700A/B</td>
<td>Installation of a Rockwell Collins TCAS-400 Traffic Alert and Collision Avoidance System (TCAS-II).</td>
<td>The Certification Basis for the original product as amended by the following additional alters. The Certification Basis for the original product is as amended by the following additional alter(s): The certification basis for the original product is as amended by the following additional alter(s): The certification basis for the original product is as amended by the following additional alter...</td>
<td>Environmental Protection and...</td>
<td>Environmental Protection and...</td>
<td>Environmental Protection and...</td>
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<tr>
<td>AVIATION PARTNERS BOEING WINGLET S COMPANY, LLC</td>
<td>FAA STC 015185E</td>
<td>REV. 6</td>
<td>FAA</td>
<td>757-200-300</td>
<td>Install of Aviation Partners Boeing Winglets on B757-200 and B757-300Airplanes</td>
<td>The Partial Wing Retrofit Installation (757-300 - 1 Kit only) is approved by Rolls Royce RB11-05E powered a/c. See the following Aviation Partners Boeing drawings for aircraft types and models approved by EASA/ FAA Approved Model List (AML,No. 2-22566-002, Rev. K, 12/08/2011) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed de...</td>
<td>FAA Approved Mod Kit (AML,No. 2-22566-002 Rev. K, 12/08/2011)</td>
<td>10/06/2016</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>ST03027AT-D</td>
<td>ARMSTRONG AEROSPACE Inc.</td>
<td>REV. 4</td>
<td>FAA STC ST03027AT-D</td>
<td>737-700/-800/-900/-1000/-1100/-1200</td>
<td>Installation of Honeywell RESCU 408AF/ AFN/AFN2 Emergency Locator/Transmitter (ELT) System Revision 1 extends the applicability to the models 737-700/-800/-900/-1000/-1100/-1200</td>
<td>Certification Master Data List, Doc. No. CN19-039464-L-00, Revision 0 dated 20 December 2019</td>
<td>Prior to installation of this design change/repair it must be determined that the interrelationship between this design change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/03/2020</td>
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<tr>
<td>ST02162CH</td>
<td>CARLISLE INTERCONNECT TECHNOLOGIES</td>
<td>REV. 4</td>
<td>FAA STC ST02162CH</td>
<td>737-700/-800/-900/-1000/-1100/-1200</td>
<td>Installation of Elementary or Enhanced Surveillance Mode S Transponders (ECS) - Rev. N, dated 8 April 2008</td>
<td>Certification Master Data List, Doc. No. DC104-30584-00V&lt;+1&gt;</td>
<td>Only Configurations 1, 2, 4, 5, 6, 7, 8 and 9 are deemed to be approved. Prior to installation of this design change it must be determined that the interrelationship between this design change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/07/2016</td>
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<tr>
<td>ST01773CH</td>
<td>CARLISLE INTERCONNECT TECHNOLOGIES</td>
<td>REV. 4</td>
<td>FAA STC ST01773CH</td>
<td>737-700/-800/-900/-1000/-1100/-1200</td>
<td>Installation of ICAO 4D FM Transponders (AFM)</td>
<td>Certification Master Data List, Doc. No. DC104-30584-00V&lt;+1&gt;</td>
<td>Only Configurations 1, 2, 4, 5, 6, 7, 8 and 9 are deemed to be approved. Prior to installation of this design change it must be determined that the interrelationship between this design change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/07/2016</td>
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<td>ST015664</td>
<td>GC1413083001, Révis 1</td>
<td>REV. 2</td>
<td>FAA STC ST015664</td>
<td>GIV-X</td>
<td>Installation of Broad Band Multi-Link (BBML) Systems - Revision of the EASA STC.</td>
<td>Certification Master Data List, Doc. No. CN19-039464-L-00, Revision 0 dated 20 December 2019</td>
<td>Prior to installation of this design change/repair it must be determined that the interrelationship between this design change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/06/2014</td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10015713</td>
<td>FOKKER 50</td>
<td>REV. 2</td>
<td>FAA STC ST01916A</td>
<td>FOKKER 50</td>
<td>Installation of Large Cargo door</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and are applicable to this certificate/approval.</td>
<td>1. Maintenance Manual Supplement to MMS 222 Rev. S.3, FAA approved.</td>
<td>12/02/2016</td>
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<tr>
<td>10015717</td>
<td>CASCADE AEROSPACE Inc.</td>
<td>REV. 2</td>
<td>TCCA STC SA00-69</td>
<td>DHC-8-314, DHC-8-315 DHC-8-301, DHC-8-311 DHC-8-301, DHC-8-202 DHC-8-103, DHC-8-106 DHC-8-102</td>
<td>Environmental Control System Performance Improvement Modification/Revision 01: Editorial Changes to include Issue 04 of TCDA STC SA00-69</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for additional or alternative environmental protection and the associated certificate noise and/or emissions levels of the product must be approved to this certificate/approval without any impact.</td>
<td>1. Change defined with Revision 17 of Cascade Aerospace Inc. Master Data Document MDL 2302. Maintenance shall be accomplished in accordance with Revision 04 of Cascade Aerospace Inc. maintenance manual supplement document MMS 2302 Date revisions of the above listed document(s) approved/applicable.</td>
<td>22/02/2021</td>
</tr>
<tr>
<td>10015720</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>REV. 3</td>
<td>FAA STC ST01849SE</td>
<td>777-200</td>
<td>Business Class Seat Installation</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CRI H01 EWS The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and are applicable to this certificate/approval/applicable. Corresponding maintenance manual supplement document MMS 2302 Date revisions of the above listed document(s) approved/applicable.</td>
<td>Boeing Structural Compliance summary Report Document DIN-36718 Revision New with dated 17 March 2008 must be accomplished concurrently with this STC. Boeing Structural Compliance Summary Report Document DIN-36718 Revision New dated 17 March 2008 must be accomplished concurrently with this STC. Boeing Structural Compliance Summary Report Document DIN-36718 Revision New dated 17 March 2008 must be accomplished concurrently with this STC. Concurrent Info...</td>
<td>02/02/2012</td>
</tr>
<tr>
<td>10015727</td>
<td>ROCKWELL COLLINS AEROSPACE ELECTRONICS, Inc.</td>
<td>REV. 3</td>
<td>FAA STC ST01950A</td>
<td>CL-600-2D24, CL-600-2E25, CL-600-2C10, CL-600-2D15</td>
<td>Installation of Head-Up Guidance System (HGS) Model 4200 Addition of CL-400-2E25 (CRJ 1000) to HGS 4200 STC</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CRI A-1000-01-HGS The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and are applicable to this certificate/approval/applicable.</td>
<td>-EASA STC no. EASA.IM.A.S.01743 Revision 1 dated 31 May 2007. FAA STC no. ST01950A dated 28 December 2010 Rockwell Collins Master Drawing List No. 9701-1099 at Revision N or later approved revision. Rockwell Collins Airplane Flight Manual Supplement No. 9701-1152 at Revision E or later approved revision</td>
<td>None</td>
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<tr>
<td>10015728</td>
<td>SPECTRUM AERONED, LLC/aka SPECTRUM AEROMED</td>
<td>REV. 2</td>
<td>FAA STC SA02177CH</td>
<td>JETSTREAM 3201</td>
<td>Removal of passenger seats and installation of Spectrum Aeromed mediasystems</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA02177CH. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product</td>
<td>Master Data Document 6034-2A, Revision 1, dated July 6, 2009. Spectrum Aeromed Inc. AFM supplement 6034-2A-MS/AFM/Aeronmed Instructions for Continued Airworthiness Manual 6043-1M/SR later revision of the above listed documents approved by EASA in accordance with EASA ED decision 200404/C (or subseq...)</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAACert ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>AERONAUTICAL ENGINEERS Inc.</td>
<td>REV. 4</td>
<td>FAAST01827LA</td>
<td>737-300, 737-400</td>
<td>BOEING 737-300/400 FREIGHTER WITH OPTIONAL CARGO CONTAINER/CONFIGURATION Modificat ion of notch frames to accommodate 88x125 container disposition 10 (year)</td>
<td>- AEI Cargo Modification Drawing List R-668, Revision L, dated June 21, 2010 for B737-300 aircraft. - AEI Cargo Modification Drawing List R-811, Revision E, dated June 21, 2010 for B737-400 aircraft. - AEI Inspection Instructions Document R-834, Revision A, dated September 22, 2010 or later revisions offa...</td>
<td>Limited to B737-300/400 modified with EASA STC IMAS.0174ARev 3 (FAA/ST01827LA)</td>
<td>24/02/2010</td>
<td>Active</td>
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<tr>
<td>DAYTON GRANGERS Inc.</td>
<td>REV. 5</td>
<td>FAA STC NO. ST0241NY</td>
<td>CL-600-2E25 (RJ SERIES 1000) CL-600-2D24 (RJ SERIES 900) CL-600-2D15 (RJ SERIES 700) CL-600-2C10 (RJ 700/701/702) CL-600-2B10 (RJ SERIES 100)</td>
<td>Installation of Winglet and Horizontal Stabilizer End Cap Lightning Protection Kits</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>3/10/2016</td>
<td>Active</td>
</tr>
<tr>
<td>FIELD AVIATION EAST, LTD.</td>
<td>REV. 3</td>
<td>SA4A-125 ISSUE 5</td>
<td>DHC-8-315 DHC-8-314 DHC-8-311 DHC-8-301 DHC-8-202 DHC-8-201 DHC-8-106 DHC-8-103 DHC-8-102</td>
<td>Retrofit of Collins Elementary Surveillance and ADSB capable TDR-94D, Dual Mode-S Transponders and associated Collins CTL92E, or Universal RCU Controllers. Revision 3 incorporates ADSB capability via Configuration C.</td>
<td>Configuration A, Elementary Surveillance Capable Transponders with CTL92E. Configuration D, ADSB Capable Transponders with CTL92E or Universal RCU Controllers. The installation of Field Aviation STC SA89-72, TCAS II/WACAS II and Dual Mode-S Transponders or the fitment of Bombardier Factory...</td>
<td>Configuration A, Elementary Surveillance Capable Transponders with CTL92E Controller and Configuration B, ADSB Capable Transponders with CTL92E or Universal RCU Controllers. The installation of Field Aviation STC SA89-72, TCAS II/WACAS II and Dual Mode-S Transponders or the fitment of Bombardier Factory...</td>
<td>6/06/2019</td>
<td>Active</td>
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<tr>
<td>ACORN WELDING Ltd.</td>
<td>REV. 1</td>
<td>TCCA STC SA4A-63</td>
<td>182R 182L/182M/182M/182N/182P/182Q 182E/182F/182G/182H/182J/182K</td>
<td>Installation of Aerocet 3400 Amphibious Float Conversion.</td>
<td>In accordance with Aerocet Aeronautical Limited (AAL) Installation Instructions EO333901 dated May 10, 2004, and Master Data List MSL3301 Rev 2 dated November 10, 2004, or later approved revisions. TCCA approved Flight Manual Supplement FMS330962 dated May 10, 2004 is required for Cessna Models 182E...</td>
<td>1. This STC confers no Operational Approval; that must be sought from the National Airworthiness Authority of the country of registration. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility...</td>
<td>14/06/2017</td>
<td>Active</td>
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<tr>
<td>AEROLITE AG</td>
<td>REV. 3</td>
<td>35A</td>
<td>EMS Equipment Installation Cabin installation of Patient Unit PN 250075-509, Power Box PN250080B-501 and Cabinet PN 250073-604</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- According to Aerolite Master Document List No.MDL-06-001 Rev C-To be operated in accordance with Aerolite AFA-5 Document No.FMS-06-001, Rev B-To be maintained i.e. Aerolite/CA Document No. OPM-68-001, Rev C or later revisions of the above listed document(s) approved/ accepted/understand by the EASA sys...</td>
<td>Limited to: SW 626, 663, 685 Prior to installation of this change/repair it must be determined whether the new/interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>7/08/2019</td>
<td>Active</td>
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<td>Issuing Agency</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>STC No.</td>
<td>FAA STC No.</td>
<td>STC Type</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
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<td>STC Holder</td>
<td>FAA STC</td>
<td>Rev.</td>
<td>Foreign/NAA STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>JET AVIATION AG</td>
<td>REV. 3</td>
<td>A320-271N/-272N/-273N</td>
<td>A320-251N/-252N/-253N</td>
<td>A320-233</td>
<td>A320-231/-232/233</td>
<td>A320-212/214/215/-215/-216</td>
<td>A320-211/-212</td>
<td>A320-212/-214/-215/-216</td>
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<td>AEROMECH Inc.</td>
<td>REV. 1</td>
<td>FAA SA00588E-D</td>
<td>B300C</td>
<td>B200/CT/B200/</td>
<td>300/300L/B300/</td>
<td>200/200C/300CT&amp;B200/</td>
<td>B200C</td>
<td>Installation of a Pilot and Co-Pilot Digital Air Data Unit</td>
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<tr>
<td>ARMSTRONG AEROSPACE Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST0182CH</td>
<td>737-800, 737-900, 737-900ER</td>
<td>737-700, 737-700, 737-700</td>
<td>737-700, 737-700</td>
<td>737-700, 737-700</td>
<td>737-300, 737-400, 737-200</td>
<td>Installation of a Honeywell RESCU 406AF/AFN/AFN2 Emergency Locator Transmitter (ELT) System.</td>
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<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA ST0012AT</td>
<td>737-300, 737-400, 737-500</td>
<td>737-300, 737-400, 737-500</td>
<td>737-300, 737-400, 737-500</td>
<td>737-300, 737-400, 737-500</td>
<td>737-300, 737-400, 737-500</td>
<td>Installation of an Active Noise Reduction (ANR) Headset. Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
</tr>
<tr>
<td>WIPARE Inc.</td>
<td>REV. 1</td>
<td>FAA STC SA02CH</td>
<td>DHC-6 SERIES 400</td>
<td>DHC-6 SERIES 300</td>
<td>DHC-6 SERIES 400</td>
<td>DHC-6 SERIES 300</td>
<td>DHC-6 SERIES 400</td>
<td>Installation of Winglet 13000 Amphibious Floats or Installation of Winglet 13000 Seaplane Floats on DeHaviland DHC-6 Series 300 and 400.</td>
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### List of EASA Supplemental Type Certificates

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<th>STC no.</th>
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<th>Rev.</th>
<th>Description</th>
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<th>Associated Technical Documentation</th>
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<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>ST09262SC-D</td>
<td>WIPAIRE INC.</td>
<td>REV. 3</td>
<td>Installation of an Amphibian Gear Positioning System and Laser Gear Advisory System in accordance with Wipaire Inc. Master Drawing List W- MD151, Rev. C and the drawings listed on AIMs. To this STCor later FAA approved revisions.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CB 23.305(a), 23.305(b), 23.601, 23.603, 23.607, 23.609, 23.611, 23.613, 23.720(b)(23.617), 23.1301(a), 23.1322, 23.1351(a)</td>
<td>FAA Parts 42, 14, 255-105-01, ICC-105</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/01/2016</td>
<td>Active</td>
<td>10015832</td>
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<tr>
<td>ST09901AT</td>
<td>A.A.E., LTD.</td>
<td>REV. 1</td>
<td>Installation of an AAE, List smoke detection and halon suppressionsystem into the lower baggage compartments of theaircraft to establish class C cargo compartment classification. The installation includes a control head installed in the aft equipment bay for monitoring and operation of the system...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Parts 42, 14, 255-105-01, ICC-105</td>
<td>Refer to the limitation section of AAE, Ltd Airplane Flight Manual Supplement. No. TF06-0087, Rev. IR, dated January 14, 2005 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU.</td>
<td>16/11/2015</td>
<td>Active</td>
<td>10015840</td>
</tr>
<tr>
<td>SA05-42</td>
<td>TCCA</td>
<td>BD-100-1A10</td>
<td>Installation of an ARTEX ELT t-406 with a navigation interface Revision of the EASA STC T.C.A.S.01938 REV 1 in order to delete the structural provisions approved under STC TCCA SA-06-29 as a prerequisite to TCCA STA-06-14 for future air deliveries Rev. 3 removes the commercial designation from the...</td>
<td>Certification Basis as per EASA TCDS IM.A.080 plus: The TCCDS IB.080 or later certification basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STA-06-14, Issue 2, dated 09th September 2010; Modification Summary E618000 Revision 0 dated 10th September 2009; Compliance Check List E618066 Revision D dated 29th August 2010; for Continued Airworthiness RAL-83/105-2550-I-01A Revision...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/04/2018</td>
<td>Active</td>
<td>10015843</td>
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<tr>
<td>ST09901AT</td>
<td>FIELD AVIATION EAST LTD.</td>
<td>REV. 3</td>
<td>Installation of Dual Universal GLS-1250 DGPS Receivers (SCAT-1) (TCCStc SA05-42, Issue 4) Revision 3 was issued to include Configuration C</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Parts 42, 14, 255-105-01, ICC-105</td>
<td>For Configuration A and Configuration B: FAA Parts 415-1F, 155-1F, 425-1F, 155-1F, 255-1F, 155-1F, 435-1F, 155-1F, 255-1F with appropriate software revisions 801/802 and 803/STC EASA IM.A.S.3183, Issue 1 or later FAA/STC approved revision must be installed as a prerequisite to the installation of the Dual GLS-1250 DGPS (SCAT-1) modification. For Configuration C: FAA Parts 425-1F, 255-1F, 155-1F, 435-1F, 155-1F, 255-1F with appropriate software revisions 801/802 and 803/STC EASA IM.A.S.3183, Issue 1 or later FAA/STC approved revision must be installed as a prerequisite to the installation of the Dual GLS-1250 DGPS (SCAT-1) modification.</td>
<td>08/11/2019</td>
<td>Active</td>
<td>10015849</td>
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<td>ST09901AT</td>
<td>BEECHCRAFT CORPORATION</td>
<td>REV. 3</td>
<td>Installation of a Third and Fourth 750 Liter Oxygen Bottle in ATR equipment bay...</td>
<td>FAA Parts 42, 14, 255-105-01, ICC-105</td>
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<td>18/02/2010</td>
<td>Active</td>
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<td>BEECHCRAFT CORPORATION</td>
<td>REV. 3</td>
<td>FAA ST10011SC-0</td>
<td></td>
<td></td>
<td>Installation of a Honeywell Solid State Flight Data Recorder with interface to</td>
<td>FAA CAV 46 &amp; 14 CFR Part 25 The Certification Basis for the original product remains applicable thtts certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this c...</td>
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<tr>
<td>GENERAL AVIATION SERVICE Aps</td>
<td>REV. 1</td>
<td>FAA SA654N-0</td>
<td></td>
<td></td>
<td>Installation of a Textron Lomong Model C-365-44, A44 or A4N engine and Searonch Model 762E6914-0-40 fixed Pitch metal propeler.</td>
<td></td>
<td></td>
<td></td>
<td>06/04/2020</td>
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<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC 70268CH-0</td>
<td></td>
<td></td>
<td>Installation of Rosemount Aerospace Door Obstacle Surveillance System (CDSS).</td>
<td>The Certification Basis for the original product remains applicable thtts certificate/ approval.</td>
<td></td>
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<tr>
<td>ISRAEL AEROSPACE INDUSTRIES, LTD.</td>
<td>REV. 2</td>
<td>FAA STC 70268N-0</td>
<td></td>
<td></td>
<td>Conversion of Boeing 747-400 Aircraft from Combi to Special Firefighter/Configuration.</td>
<td>- Master Document List Number 366-0-04-0550 Revision &quot;New&quot; or later FAA approved revision.</td>
<td></td>
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<tr>
<td>BLR AEROSPACE, L.L.C.</td>
<td>REV. 4</td>
<td>FAA SA15155E-0</td>
<td></td>
<td></td>
<td>Installation of winglets and associated changes in wing/landing gear structures and related system changes on Raytheon Model 200 200C, B200, 200C, 300 and 300L aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
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<tr>
<td>10015951</td>
<td>B/E AEROSPACE, Inc.</td>
<td>JETSTREAM SERIES 3200</td>
<td>Installation of a Honeywell CAS 67A Traffic Alert and Collision Avoidance System (TCAS IT) and Dual Mode MST 67A Mode S Transponder</td>
<td>Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
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<td>02/02/2016</td>
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<td>10101501</td>
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<td>10015956</td>
<td>BOMBARDIER Inc.</td>
<td>TCASA06-37</td>
<td>Installation, Structural Provisions Security Syst. Antenna</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<td>17/09/2021</td>
<td>Active</td>
<td>10101596</td>
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<tr>
<td>10015977</td>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>DHC 6-100, 6-200, 6-300</td>
<td>Installation of Single or Dual Sandel ST3400 Terrain Awareness Warning System (TAWS/RMI) System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
<td></td>
<td>17/03/2010</td>
<td>Active</td>
<td>10101597</td>
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<tr>
<td>10015989</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>GIV-K, GV-SP</td>
<td>Installation of Securaplane Security System in Gulfstream models GIV-Kand GV-SP</td>
<td></td>
<td></td>
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<td>24/02/2011</td>
<td>Active</td>
<td>10101589</td>
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<tr>
<td>10016001</td>
<td>GOODRICH INTERIORS</td>
<td>B767-200-300</td>
<td>Installation of “ Sentry One” Flight Deck Door Camera Surveillance System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
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<td>16/02/2012</td>
<td>Active</td>
<td>10101601</td>
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<td>STC Holder</td>
<td>Rev.</td>
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<td>Models</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST02794AT-D</td>
<td>GIV-SP</td>
<td>Installation of Board Band Multi-Link System (BBML). Revision of the EASA STC EASA.IM.A.S.02167</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>None</td>
<td>20/06/2012</td>
<td>Active</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 3</td>
<td>FAA STC ST02338AT-D</td>
<td>GIV-X / GIV-SP</td>
<td>Installation of Cabin Communication System. Revision 1: Editorial changes in Associated Technical Documentation and Description fields. New STC number (previous number was IM.A.S. 02174). Revision 2: Editorial changes in Model applicability section. Revision 3: Editorial changes in Applicant address and T...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Rev...</td>
<td></td>
<td>Prior to installation of this design change it must be determined if there is an detrimental interaction between this design change and any other previously installed design change and/or repair will introduce roadwreck effect upon the airworthiness of the product.</td>
<td>27/06/2016</td>
<td>Active</td>
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<tr>
<td>BOMBARDIER Inc.</td>
<td>REV. 66</td>
<td>TCCA SA08-90</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Installation of Complete Customized Aircraft Interior.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): Transport Canada SCA 2022-08 Non-Rechargeable Lithium Batteries. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>See those of TCCA STC SA 08-60. The EASA approved configurations are identified in the Configuration/Definition List document Ref E934000/aircraft issued with a CoA on or after 18 May 2019 must be equipped with Halon Free extinguishers.</td>
<td>24/10/2022</td>
<td>Active</td>
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<tr>
<td>WHELEN AEROSPACE TECHNOLOGIES LLC</td>
<td>REV. 7</td>
<td>FAA ST02893AT</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of a Boom Beam Taxi/Landing Light System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List Report No. 47: rev. A. For later revisions of the above listed document(s) accepted on behalf of EASA in accordance with Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>The STC is valid for aircraft types and models approved by EASA and listed on the latest FAA approved Flight Manual Supplement as applicable.</td>
<td>26/03/2021</td>
<td>Active</td>
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<tr>
<td>AEROMECH Inc.</td>
<td>REV. 3</td>
<td>FAA STC ST01526SE</td>
<td>SS50 (CITATION SII) 560 (CITATION V) 550 (CITATION I) 560 (CITATION I)</td>
<td>Group Aircraft Approval for Reduced Vertical Separation Minimum (RVSM)</td>
<td>The EASA TCDS IM.A.207 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. 1) Applicant Technical Documentation (Aeromech) - AMI-QD-C5-007, Rev F; Instructions for Initial and Continued Airworthiness: Depending on the class of installed avionics on the aircraft: AMI-ICA-C5-009: for Class 1 airframes (group aircraft equipped with Shad in ADC 6000 on data computers) none</td>
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<td>08/08/2011</td>
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<td>LB AIRCRAFT ENGINEERING LLC</td>
<td>REV. 2</td>
<td>FAA ST03364AT</td>
<td>737-300-410 737-300</td>
<td>Installation of a single Universal Avionics Class &quot;A&quot; TAWS system/Revision 1: Transfer: STC 10016083 (EASA.IM.A.S. 02267) from ARC AvionicsCorporation to AIRCRAFT RADIO &amp; AVIONICS, LLC/Revision 2: STC Transfer from &quot;Aircraft Radio &amp; Avionics, LLC&quot; to LB AIRCRAFT Engineering, LLC.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>-ARC Avionics Master Data List 762-061505, Rev L, dated 5 April 2008 &amp;/or in later revisions of the above listed documents approved by EASA.</td>
<td>29/02/2024</td>
<td>Active</td>
<td>1016083</td>
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<tr>
<td>LB AIRCRAFT ENGINEERING LLC</td>
<td>REV. 2</td>
<td>FAA ST03365AT</td>
<td>737-300</td>
<td>Installation of a dual Universal Avionics UMS-1F EMS system/737-300 aircraft. Installation includes 2 Nav Control Units (NCU) in the EEE bay and 2 GPS antennas. Center pedestal pick and co-pilot Control Display Units (CDU)/Revision 1: Transfer: STC 10016084 (EASA.IM.A.S.02269) from ARC AvionicsCor...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>-ARC Avionics Master Data List 750-120605, rev L, dated 8 April 2008 - Flight Manual Supplement Doc 730-061505, Rev B, dated 25 April 2008 &amp;/or in later revisions of the above listed documents approved by EASA.</td>
<td>29/02/2024</td>
<td>Active</td>
<td>1016084</td>
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<tr>
<td>LB AIRCRAFT ENGINEERING LLC</td>
<td>REV. 4</td>
<td>FAA ST03362AT</td>
<td>737-200-370 737-200-350</td>
<td>Installation of a single Universal Avionics UMS-1F Flight Management System (FMS), or UMS-1f WAAIS capable FMS, for Boeing 737-200 and 737-300 aircraft with original Honeywell or Collins EFIS/Configuration II/Installation includes: 2 Nav Contr...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (a) (CR): H-01 Issue 3 dated 07 March 2011; Special condition - EnhancedAirworthiness Program for Aeroplane Systems – ICA on EWL/ The requirements for env...</td>
<td>Master Data List: MDL 752-120605 Rev C, dated 02 April 2008 (B737-300 and B737-300 A/C without EFIS)- MDL 752-022310 Rev E, dated 22 Jan 2020 (B737-300 A/C with EFIS)/ Flight Manual Supplements- AFMS 732-022310 Rev B dated 28 Jun 2013 (B737-300 A/C with EFIS)/ AFMS 732-022310 Rev B dated 07 May 2009 (B7)</td>
<td>29/02/2024</td>
<td>Active</td>
<td>1016085</td>
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<tr>
<td>AIRCRAFT RADIO &amp; AVIONICS, LLC</td>
<td>REV. 1</td>
<td>FAA ST03363AT</td>
<td>737-200</td>
<td>Installation of a single Universal Avionics Class &quot;A&quot; TAWS system/Revision 1: Transfer: STC 10016088 (EASA.IM.A.S.02271) from ARC AvionicsCorporation to AIRCRAFT RADIO &amp; AVIONICS, LLC/Revision 2: STC Transfer from &quot;Aircraft Radio &amp; Avionics, LLC&quot; to AIRCRAFT RADIO &amp; AVIONICS, LLC.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>-ARC Avionics Master Data List 762-081505, Rev -LR, dated 08 February 2007 (B737-200 &amp; B737-300 A/ C with EFIS)/ Flight Manual Supplement Doc 732-081505 Rev A, dated 08 February 2007/ or subsequent revisions of the above listed documents approved by EASA.</td>
<td>29/09/2017</td>
<td>Active</td>
<td>1016086</td>
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<td>AIR ALLIANCE GmbH</td>
<td>REV. 3</td>
<td>PA-42-720</td>
<td></td>
<td>Extension of pressurized fuselage life limit to account for changes in operational usage from Alitalia Airlines Training to Normal Usage by operator: Note: This STC superseded STC EASA.IM.A.S. 02278 Rev 2 and indicates the transfer of applicable S/N to a new STC holder by Air Alliance.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. FAR 23.401(k) to 1b The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Air Alliance GmbH Report No AARD-1-Rev A dated 28 Dec 2009 &amp; Air Alliance PA-42-720 Aircraft - Fuselage Life Substantiation Air Alliance GmbH Report No AARD-2 Rev B dated 28 Dec 2006 &amp; Air Alliance PA-42-720 Aircraft - Pressurized Fuselage Cabin Installation/Continued Airworthiness/Air Alliance MDLo...</td>
<td>19/07/2011</td>
<td>Active</td>
<td>1016091</td>
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<td>PRECISE FLIGHT, Inc.</td>
<td>10016009</td>
<td>REV. 2</td>
<td>FAA STD 10108E</td>
<td>SR22, SR22T</td>
<td>Installation of Cirrus Fixed Oxygen System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. CS 23. EASA certification basis defined in TCDS IM A.007 ref CRI A-01. This certificate/approval involves a change to the certification basis.</td>
<td></td>
<td></td>
<td>03/08/2010</td>
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<td>TELEHYDE CONTROLS, LLC</td>
<td>10016095</td>
<td>REV. 3</td>
<td>FAA STC ST01547LA</td>
<td>BD-700-1A10</td>
<td>Installation of EMS Technologies High Speed Data Satellite Communication System HSD-64 and HSD-128 and validation of FAAST STS194TLA. This STC supersedes EASA-IM.A.S.02294. Following a change in the applicable text.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environment/prot and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>12/04/2016</td>
<td>Active</td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>10016111</td>
<td>REV. 1</td>
<td>FAA ST03345AT</td>
<td>767-200, 767-300</td>
<td>Installation of an Enhanced Surveillance System (EHS) to expand Transponder Mode-S functionality. Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None. Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10161111</td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>10016113</td>
<td>REV. 1</td>
<td>FAA ST0297AT</td>
<td>A300-84-103</td>
<td>This installation added three additional parameters to be recorded by the existing Flight Data Recorder. Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10161113</td>
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<tr>
<td>ANORA INTERNATIONAL LLC</td>
<td>10016114</td>
<td>REV. 1</td>
<td>FAA ST01690A</td>
<td>737-400 SERIES</td>
<td>Installation of cargo loading system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. STCs EASA-IM.A.S.02438 (FAA SAA30800) and EASA-IM.A.S.02440 (FAA SAA30700) must be installed in conjunction with this installation. 2. Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>18/03/2016</td>
<td>Active</td>
<td>10161114</td>
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<td>FLYHT AEROSPACE SOLUTIONS Ltd.</td>
<td>10016123</td>
<td>REV. 1</td>
<td>TCCA STC SA08-33</td>
<td>A321-111/-121/-211 A320-211/-212/-214/231/-232 A319/-111/-121/-131/-114 AFIRS 220/228 Automated Flight Information Reporting System Installation with Optional Global Voice and Global Text Satellite Communications</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-4AFIRS 220 system - Master Document List, FLYHT Aerospace List, Flight Data Recorder Instructions for Continued Airworthiness. FLYHT-Aerospace document No. 153-0025 Revision E dated 10 April 2013. AFIRS 228 system - Master Document List, FLYHT Aerospace.1...</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>18/11/2014</td>
<td>Active</td>
<td>10161213</td>
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List of EASA Supplemental Type Certificates

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<tr>
<th>EASA STC No.</th>
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<th>Rev.</th>
<th>Foreign/NAACertificate</th>
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<tr>
<td>10016129</td>
<td>FIELD AVIATION EAST, LTD.</td>
<td>REV. 5</td>
<td>TCCA STC: SA07-80</td>
<td>DHC-8-311</td>
<td>Retrofit of Swedish Coastguard Maritime Partol Modifications. Note: Initial EASA Supplemental Certificate was issued with Certificate Number: EASA.IM.A.S. 02323 Revision 01.</td>
<td>Configuration/installation must be performed in accordance with FABL Master Drawings Data Lists designated DDL-1141 Rev H-I dated September 28, 2021 Flight Operations must be performed in accordance with FABL FlightManual Supplements as followed AMS 2015 Rev B EGPRS Override Contr...</td>
<td>Restricted to ASN 9016, 9020, 9036, 9054, 9098, 9109, 9049, 9023. The following STC's are a concurrent requirement with this STC - EASA STC 10001655 (FAA STC ST00941IA) - 29/09/2015</td>
<td>07/06/2023</td>
<td>Active</td>
<td>10016129</td>
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<td>10016140</td>
<td>LEARJET Inc.</td>
<td>REV. 6</td>
<td>FAA SAI164M-1D</td>
<td>BD-700-1A10</td>
<td>Installation of a Complete Custom Aircraft Interior Fix: 5. Addition of ACSN 9949 and an update to the holder's address Rev. 6: Addition of ACSN 9023.</td>
<td>Certification Basis for the original product is applicable to this certificate/ or repair will introduce an adverse effect upon the airworthiness of the aircraft.</td>
<td></td>
<td>Restricted to ABN 9116, 9020, 9036, 9054, 9098, 9109, 9049, 9023. The following STC's are a concurrent requirement with this STC - EASA STC 10001655 (FAA STC ST00941IA) - 09/10/2023</td>
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<td>10016155</td>
<td>VISIONSAFE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST00731A</td>
<td>767-200-300-300F</td>
<td>Installation of Emergency Vision Assurance System (EVAS) Revision 1: Model B767-300F added</td>
<td>Certification Basis for the original product is applicable to this certificate/ or repair will introduce an adverse effect upon the airworthiness of the aircraft.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other subsequently installed design change and/or repair will introduce a new adverse effect upon the airworthiness of the aircraft product.</td>
<td>29/06/2015</td>
<td>Active</td>
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<td>10016174</td>
<td>TELEDYNE CONTROLS, LLC</td>
<td>REV. 2</td>
<td>FAA STC ST00334A</td>
<td>747-400</td>
<td>Installation of Aircraft Communications, Addressing and Reporting Systems (ACARS): This STC supersedes EASA.IM.A.S. 02401 following a change in the applicable address.</td>
<td>Certification Basis for the original product remains applicable to this certificate/ or repair will introduce an adverse effect upon the airworthiness of the aircraft.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other subsequently installed design change and/or repair will introduce a new adverse effect upon the airworthiness of the aircraft product.</td>
<td>12/04/2016</td>
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<td>10016180</td>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST00394AT</td>
<td>JETSTREAM 4100</td>
<td>Installation of an Avionics Innovations DMP-100-200 Digital Media Player (DMP) for the reproduction of Boarding Music and Cabin Briefings. Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>Certification Basis (CS) for the original product remains applicable to this certificate/ or repair will introduce an adverse effect upon the airworthiness of the aircraft.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other subsequently installed design change and/or repair will introduce a new adverse effect upon the airworthiness of the aircraft product.</td>
<td>02/03/2016</td>
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<td>STC Holder</td>
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<td>SAFRAN CABIN INC.</td>
<td>B777-200</td>
<td>Pantry installation PN C801001-101, PN C801001-103 and C802001-101</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Drawing list C802000 rev. 4 dated April 22, 2009 (i.e., 99 Y-Zodiac drawing list number C802000 Rev. C dated October 20, 2009) FAA amended STC No. ST20144LA: Later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04SC-C (or subsequent revisions of this decision).</td>
<td>Prior to installation of this design change it must be determined that there is no adverse interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/11/2013</td>
<td>Active</td>
<td>1011210</td>
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<td>DASSAULT FALCON JET CORPORATION - COMPLETION CENTER - (HANGAR 2 AND 3)</td>
<td>MYSTERE FALCON 50 (MF50)</td>
<td>Installation of the Inertial Reference System Modifications/Installation of Securaplane XL-245 batteries in lieu of Saft batteries (provide backup power) in accordance with Drawing List report No. F360-01168 Revision NC dated 23rd July 1999 for later approved revisions.</td>
<td>FAA STC ST93045SC-D: Airplane Maintenance Manual Supplement Report No. 39606 (SecuraplaneXL245 Battery) dated 30th June 1999 or later approved revisions.</td>
<td>As per FAA STC ST93045SC-D</td>
<td>04/05/2012</td>
<td>Active</td>
<td>1011215</td>
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<tr>
<td>CHEYENNE TECH</td>
<td>PA-43-700 C801001-101, PA-43-7000 C801000-101</td>
<td>Dual Digital Air Data System Installation Qualified for Operation inflightSM Airspace installation number 4 for equipment configuration: Dual Garmin G800 integrated display system, AL-285 Altitude Alerter, Skylight ADC-202AAU, APS-65 FD/AP.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement FNS800-34245, R/7 September 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04SC-C (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EASA/Blair Agreement.</td>
<td>This revision 1 is limited to Installation 4 as per MDL.800-34245, Rev.5/15 August 2011 for equipment configuration: Dual Garmin G800 integrated display system, AL-285 Altitude Alerter, Skylight ADC-202AAU, APS-65 FD/AP. Prior to installation of this design change it must be determined that there is no adverse effect on the airworthiness of the product.</td>
<td>14/03/2016</td>
<td>Active</td>
<td>1011223</td>
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<tr>
<td>ISRAEL AVIATION INDUSTRIES LTD IA</td>
<td>B737-400</td>
<td>Major Change to STC 10016227 Revision 4 for the ILJ 737-400 conversion freighter aircraft. Update of the MDL TR 365-00-00-B4200 from issue No. Issue T/All design changes being of level 2 and approved by the CAA, the rework/arrangement (EASA/CAA 22/Nov/2004) applies.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>MDL TR 365-00-00-B4200 Issue T/last revisions of the above listed documents approved by EASA.</td>
<td>Not Applicable</td>
<td>27/02/2012</td>
<td>Active</td>
<td>1011227</td>
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<td>BOMBARDIER INC.</td>
<td>TCCA STC SA07-117</td>
<td>Installation of Class 2 Electronic Flight Bag (EFB).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EASA STC No. EASA.IM.A.02508 (10016236) Rev. 2 dated 24 September 2009 TCCA STC No. SA07-117 Issue 6 dated 30 July 2009 TCCA SoC reference C-10-0435 dated 04 May 2010 Aero Consulting Services Modification Data Summary Number F376007Revision M dated 28 April 2010.Bombardier Airplane Flight Manual Sup.</td>
<td>This installation approval does not constitute operational approval of the EFB. This approval is for the EFB system structural mounting and interconnect wiring only.</td>
<td>17/02/2011</td>
<td>Active</td>
<td>1011228</td>
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<td>DECRANE AEROSPACE PATS AIRCRAFT LLC</td>
<td>B737-700</td>
<td>Installation of Canadian Marconi OMA-2101 High Gain SATCOM Antenna This STC revises and supersedes EASA STC EASA.IM.A.02510. The revisions issued to reflect the transfer from Hollingsead International to Decrane Aerospace PATS Aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Hollingsead Master Data List 12094 Rev. C dated July 07. 159690 later revision of the above document approved by EASA in accordance with EASA ED decision 2004/04SC-C (or subsequent revisions of this decision).</td>
<td>None.</td>
<td>18/11/2009</td>
<td>Active</td>
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<td>EASA STC No.</td>
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<td>Limitations/Conditions</td>
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<td>System Record</td>
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<td>01016246</td>
<td>RAABECK ENGINEERING Inc.</td>
<td>REV. 4</td>
<td>FAA STC SA3593NM</td>
<td>2 C90A, C90GT, C90GTi</td>
<td>Installation of RAABECK/Hartziei Quiet Turbofan Propeller Swiip/Blade Turbofan Propellers and optional Increased Gross Weight inaccordance with RAABECK Engineering Master Drawing List 86-2500.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certification/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>Installation of RAABECK/Hartziei Quiet Turbofan Propeller Swiip/Blade Turbofan Propellers, and optional Increased Gross Weight inaccordance with RAABECK Engineering Master Drawing List 86-2500,Revision AM, dated October 13. 2009.Aircraft must be operated in accordance with the applicable FAA STC.</td>
<td>03/07/2024</td>
<td>Active</td>
<td>1016246</td>
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<td>01016260</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST02541CH-D</td>
<td>CL-600-2B16/604 (VARIANT)</td>
<td>Installation of a Cabin Interior Arrangement, Passenger Oxygen System, and Emergency Lighting.Rev. 1: Editorial changes including name change and Major Level 2changes from FAA STC.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Jet Aviation St. Louis, Inc. Index List 41-2990G04, Revision G, dated June 2011 or later FAA approved revision. 2. FAA approved Airplane Flight Manual Supplements 41-24242/001, revision dated April 2008 applicable to configurations 1, 3, 4, 5, 6, &amp; 9, 41-2614002, Revision A, dated 10 April...</td>
<td>Restricted to ASN 5701 and subsequent aircraft.Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product...</td>
<td>31/03/2017</td>
<td>Active</td>
<td>1016260</td>
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<td>01016263</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST103166C-D</td>
<td>GULFSTREAM G150</td>
<td>Installation of an Executive cabin interior in accordance with Gulfstream Index List GA322048000.Reason for revision 1: The Satcom system is now an optional installation, updated Limitation and Conditions section, Instruction for continued airworthiness more detailed and STC holder addresschanged.</td>
<td>The Certification Basis for the original product as amended by adding additional or alternative airworthiness requirements for the Special Condition(s):D-03. Side Facing Single Occupant Seat Equivalent Safety Finding(s):D-04: Frangible Lavatory Door D-05. Emergency Exit Marker, Locator beacon...</td>
<td>Guelfstream Index List GA322048000, Rev.&quot;, dated 14 Aug 2008, Flight Manual Supplement GA321044000, Rev J , dated 12 Feb 2008, Instructions for Continued Airworthiness GA322044A20 2. Rev &quot;B&quot; dated August 7, 2006 (101 Configuration) GA32204A2 06, Rev &quot;A&quot; dated September 26, 2006 (-111 Configuration)OA32...</td>
<td>Installation of this STC requires incorporation of 1) When optional ICG/Iridium Satcom is installed, ICG/Iridium SatcomArkenna is required per STC EASA.IM.A.S.02485 (ST101345C-D)(-&gt;)2) When optional XM radio system is installed, XM antenna installations required per STC EASA.IM.A.S.02635 (ST1031...</td>
<td>12/07/2017</td>
<td>Active</td>
<td>1016263</td>
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<tr>
<td>01016292</td>
<td>SAFRAN CABIN GALLEYS US, INC.</td>
<td>REV. 1</td>
<td>FAA STC ST00943A</td>
<td>BOEING 757-200 SERIES</td>
<td>Galley installation.</td>
<td>None.</td>
<td>None.</td>
<td>None.</td>
<td>18/02/2010</td>
<td>Active</td>
<td>1016292</td>
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<tr>
<td>01016295</td>
<td>STG AEROSPACE Inc.AKA SAFTLGQ, Inc.</td>
<td>REV. 2</td>
<td>FAA STC ST03485AT</td>
<td>B737-700/-800/-900 B737-300/-400/-500</td>
<td>Installation of an STG Aerospace PowerSmart Wireless Emergency PrimaryPower System (WEPS) in accordance with STG Master Drawing List (MDL) 71430000.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval.</td>
<td>-Master Drawing List (MDL) STG Doc. No. 71430000. Rev. G dated October 5th. -2009-Instructions for Continued Airworthiness (ICA) STG Doc. No. 7143000008. B dated August 5, 2009-Enhanced Zone Analysis Procedure (EZAP) Ref. 07/13/500 Rev. J. FAA STC ST03485AT as amended July 29, 2010 for later rev...</td>
<td>Restricted to ASN 9237.This does not constitute an operational approval of the Wireless LocalArea Network (WLAN) system. It is the operator's responsibility to obtain operational approval for the airborne use of transmitting/portable electronic devices in accordance with the operating rules affr...</td>
<td>08/08/2011</td>
<td>Active</td>
<td>1016295</td>
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<tr>
<td>01016304</td>
<td>BOMBARDIER INC.</td>
<td>REV. 1</td>
<td>TCCA STC C-LSA07-285D</td>
<td>BD-700-1A10</td>
<td>Aircraft Completion, 14 Passenger Interior Configuration.</td>
<td>EASA.IM.A.S.009 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>EASA STC No. EASA.IM.A.S.00201 TCCA STC No. C-LSA07-285D Issue 1 dated 02 November 2007 TCCA SoC ref. C-10-0429 dated 01 June 2010.Aero Consulting Services Modification Data Summary F61000 Revision As dated 27 April 2010.Aero Consulting Services Flight Manual Supplement G- FM2500100 Revision/NC or later...</td>
<td>Restricted to ASN 9237. This does not constitute an operational approval of the Wireless LocalArea Network (WLAN) system. It is the operator's responsibility to obtain operational approval for the airborne use of transmitting/portable electronic devices in accordance with the operating rules affr...</td>
<td>07/06/2010</td>
<td>Active</td>
<td>1016304</td>
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[Source: easa.europa.eu]
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<th>STC Holder</th>
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<th>Rev.</th>
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<tr>
<td>BOMBARDIER INC.-COMPLETION CENTREMONTRÉ AL</td>
<td>10016316</td>
<td>REV.7</td>
<td>TCCA STO-SAO7-119</td>
<td>B760 CHALLENGER 300</td>
<td>Installation of In-Service Improvement Modifications to cabin interior features:Revision of the EASA STC 10016316 Rev. 6 due to revised technicaldata (Modification Data Summary F344000 Rev. H)</td>
<td>EASA TCS-IMA.080 Issue 3, 28th October 2010 The Certification Basis for the original product remains applicable to the STC certification/approval. The requirements for environmental protection and the associated/associated noise and/or emissions levels of the original product are unchanged and remain applicable.</td>
<td>- Transport Canada Supplemental Type Certificate SA07-119 Issue 2, dated 03rd October 2010 - Transport Canada Approved Aero Consulting Services Ltd. Modification/Data Summary F344000 Revision H dated 28th September 2010.</td>
<td>28/10/2010</td>
<td>Active</td>
<td>10016316</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>10016318</td>
<td>REV.1</td>
<td>FAA STC ST02404NY-O</td>
<td>B767-300</td>
<td>Installation of Panasonic Avionics Overhead In-Flight Entertainment System</td>
<td>The Certification Basis for the original product and the following: Additional or alternative airworthiness requirements are applicable to the certificate included in the TCDS following the following order:</td>
<td>EASA.IM.A.S.02627</td>
<td>Limited to installation on aircraft already fitted with Health/Techno/Look interior (Configuration 4) if EASA STC IMA.02474</td>
<td>11/02/2011</td>
<td>Active</td>
<td>10016318</td>
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<tr>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>10016319</td>
<td>REV.6</td>
<td>FAA STO19503E</td>
<td>777-200</td>
<td>Installation of an overhead flight crew rest above Door 1Rw 5. The new OFOIs will have B/E Lighting and Integrated Systems (WELIS) LED lighting. Panasonic Avionics (PAC) IFE components installed/compatible with the IFE system installed in the PH1 installation:</td>
<td>The Certification Basis for the original product and the following: Additional or alternative airworthiness requirements are applicable to the certificate included in the TCDS following the following order:</td>
<td>EASA.IM.A.S.02474</td>
<td>- Master Drawing List AF7S000-001 rev B/FAFM Supplement TP07-000AFM rev 2B/E-FSI Top Collector A700000-001 rev C/FAFM Supplement ST0708-D15 rev B/B-E-FSI Top Collector B761000-001 rev -</td>
<td>09/10/2020</td>
<td>Active</td>
<td>10016319</td>
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<tr>
<td>BOMBARDIER AEROSPACE</td>
<td>10016332</td>
<td>REV.1</td>
<td>TCCA C-LSA07-229/D</td>
<td>BD-700-1A11</td>
<td>Installation of Interior Compartments In-Service Improvement Modifications for ASIN 9249</td>
<td>The Certification Basis for the original product:</td>
<td>EASA STC no. EASA.IM.A.05616-TCCA STC no. C-LSA07-229/D Issue 1 dated 19th October 2007-ACS-LAI Ltd. Modification Data Summary F428000 Revision B or later approved revision: - Bombardier Airplane Flight Manual Supplement G-FAFM2500163 Revision N/ C-Instructions for Continued Airworthiness GC324660001-</td>
<td>Restricted to ASIN 9249</td>
<td>20/04/2011</td>
<td>Active</td>
<td>10016332</td>
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<tr>
<td>SPECTRUM AEROMED, LLC/ SPECTRUM AEROMED</td>
<td>10016347</td>
<td>REV.1</td>
<td>FAA STO12484CH</td>
<td>BAE 125 SERIES 1000B &amp; BAE 125 SERIES 1000A</td>
<td>Installation of an air ambulance conversion</td>
<td>The Certification Basis for the original product remains applicable to the STC certification/approval, except where amended by additional or later amendments if indicated on FAA STC STO12484CH The requirements for environmental protection and the associated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval:</td>
<td>Spectrum Aircraft Inc. Drawing 8033-66M, DS-6D, No Revision, dated January 26, 2000 or later revision of the above listed document approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this design change): - FAA Supplement dated August 4, 2007 or later approved revision.</td>
<td>Prior to installation of this design change it must be determined whether the additional design change and any other subsequently installed design change and/or repair will introduce any adverse effect upon the airworthiness of this product.</td>
<td>23/04/2019</td>
<td>Active</td>
<td>10016347</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>STC Type</td>
<td>Rev. Level</td>
<td>Foreign/NAAC</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>SPECTRUM AEROMED, LLC/a/ SPECTRUM AEROMED</td>
<td>10016349</td>
<td>FAA STC</td>
<td>REV. 1</td>
<td>SAI0882CH</td>
<td>SEE FAA AML</td>
<td>Install Air Ambulance Conversion</td>
<td>The Certification Basis for the original product remains applicable to this certificate, except where amended by additional or later amendmen...</td>
<td>Spectrum Aeromed Inc. drawings as listed on AML SAI0882CH dated 29/02/2010, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product. The STC is valid for aircraft types and models as listed on AML.</td>
<td>24/04/2019</td>
<td>Active</td>
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<tr>
<td>SPECTRUM AEROMED, LLC/a/ SPECTRUM AEROMED</td>
<td>10016350</td>
<td>FAA STC</td>
<td>REV. 1</td>
<td>SAI0972CH</td>
<td>PA-42, PA-42-1000, PA-42-720</td>
<td>Install Air Ambulance Conversion</td>
<td>The Certification Basis for the original product remains applicable to this certificate, except where amended by additional or later amendment(s) if indicated on FAA STC SAI0972CH.</td>
<td>Spectrum Aeromed Inc. Master Drawing List 6004FMS, Revision 2, dated 30/07/2012, or later FAA approved revision. - Airplane Flight Manual Supplement, 6004FMS, Revision 2, dated 22/08/2012, or later FAA approved revision. - Instructions for Continued Airworthiness, Manual 6004MM, No Revision, dated 10 J...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>24/04/2019</td>
<td>Active</td>
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<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>10016379</td>
<td>FAA STC</td>
<td>REV. 1</td>
<td>ST01431WI-D</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>EASA validation of FAA STC ST01431WI-D. Modification of a RockwellCollins, Inc. Data Base Unit (DBU) installation. In accordance with Rockwell Collins, Inc. Drawing Index RCA-0123-001 Rev G or later approved revisions. The FAA STC ST01431WI-D was initially validated by EASA on 15th October 2008. This in...</td>
<td>See those of original FAA STC ST01431WI-D as well as those of the documents listed above under associated technical documentation. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>24/04/2017</td>
<td>Active</td>
<td>10116379</td>
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<tr>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>10016381</td>
<td>FAA</td>
<td>REV. 1</td>
<td>SA10560SC</td>
<td>C90GTI</td>
<td>Installation of Wing Mounted Auxiliary Fuel Transfer Tank in BeechC90GTI. This project is to amend EASA STC No. EASA.IM.A.S.02705.</td>
<td>Installation of Wing Mounted Auxiliary Fuel Transfer Tank in accordance with Master Drawing List No. 8007-2, dated 16th August 2008. Installation of this design change must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>None</td>
<td>16/09/2011</td>
<td>Active</td>
<td>10116381</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10016375</td>
<td>LEARJET Inc.</td>
<td>REv. 1</td>
<td>FAA ST0173LA-D</td>
<td>BOMBARDIER BD-700-1A11, BOMBARDIER BD-700-1A10</td>
<td>Installation of Honeywell/EMI HS-710 High-Speed Data Salcom System. Rev.: Additional of 2 Configuration for MCS-7120 SATCOM Installation.</td>
<td>Flight Manual Supplement BAS/ST040130, Revision NC, dated 06/June 2007 Master Document List BAS/ST010038, Rev. G, dated 20/September 2023 Maintenance Instructions for Continued Airworthiness (ICA) for the MDL BAS/ST010038-1 Configuration is contained in Document BAS/ST01035B-IFCA-T-Revision NC.</td>
<td>The installation of an EASA approved Router/LAN System is a prerequisite for the installation of this modification. The installation of this modification must be certified by a TCCP of the EUR.</td>
<td>11/10/2023</td>
<td>Active</td>
<td>10101327</td>
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<tr>
<td>10016394</td>
<td>APB WINGLETS COMPANY, LLC</td>
<td>REV. 7</td>
<td>FAA STC ST0192SE</td>
<td>BOEING 767-300, -300F</td>
<td>Installation of Aviation Partners Boeing Winglets, wing stiffening reinforcement and related system changes (formerly IM.A.S.02723) Revision 4, compatibility with BCF modification as per STC 10043661 with an associated documentation update and Replacement of LED modules - PN 32-45.</td>
<td>Approved Model Lists: 767-300 and -300BCF “Partial wing Retrofit Installation” 767-300-1(K): AP67.3-AML2019 Rev2, 18th October 2018 - 767-300F “Partial Wing Retrofit Installation” 767-300F-1(K)/AP67.3F-AML2018, 3rd July 2009.</td>
<td>Operational Performance limitations are defined within the AFM. This STC is approved for aircraft equipped with General Electric CF6-80C2 series engines and Pratt and Whitney PW 4000 series engines with noise-abating structural vibration isolators installed only. The STC is approved for 67-300F airplanes.</td>
<td>10/06/2015</td>
<td>Active</td>
<td>10101334</td>
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<td>10016397</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>FAA SA1430W-00</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Modification of a Collins DBU-4000 or DBU-4100 Data Base Uninstallation to a Collins DBU-5000 Installation.</td>
<td>Installation in accordance with Collins Master Drawing List ref. AML No.EASA.IM.A.S.02726 dated, 15 October 2008, or later approved revision or later release of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and CRI.</td>
<td>1. Prior installation of Collins DBU-4000 part numbers 822-9865-001/002/003/004/005/006/007/008/009/010/011/012/013/014/015/016 via the above listed documents must be confirmed. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other modifications must be in accordance with Aircraft Flight Manual Supplement contained in Drawing No. 06/10000 Rev. B dated 27 May 2008 or later approved revision.</td>
<td>14/02/2017</td>
<td>Active</td>
<td>10101337</td>
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<tr>
<td>10016406</td>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>REV. 1</td>
<td>TCCA STC 2007-131</td>
<td>DHC-6-100, DHC-6-200, DHC-6-300</td>
<td>Installation of single or dual Sandel SA4850 primary navigation display (EHS) with optional SA4850 primary attitude display (EADP).</td>
<td>FAR 23</td>
<td>Installation in accordance with Rocky Mountain Aircraft 8(M) (Modification Instruction E100010, Rev. Transport Canada Civil Aviation (TCCA) approved March 10, 2010, or later approved revision(s) Operation in accordance with Rocky Mountain Aircraft 8(M) Flight Manual Supplement E100003, Rev. Transporter</td>
<td>None</td>
<td>03/06/2010</td>
<td>Active</td>
<td>10101400</td>
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<td>10016430</td>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>REV. 1</td>
<td>FAA SA07375C</td>
<td>B. B200CT, B200T, B. A200, A200C, A200CT, B. 200, 200C, 200CT, 200IT A. B200, 2000CT</td>
<td>Installation of Pratt and Whitney P76A-61 engines in Beach 200 Series. This project is to amend EASA STC No. EASA.IM.A.S.02732.</td>
<td>CS-23</td>
<td>Installation of Pratt &amp; Whitney P76A-61 engines in accordance with Master Drawing List No. 06/10005, Rev. B, dated 27 May 2008 or later approved revision. Operation in accordance with Aircraft Flight Manual Supplement contained in Drawing No. 06/10000 Rev. B dated 27 May 2008 or later approved revision.</td>
<td>None</td>
<td>05/07/2010</td>
<td>Active</td>
<td>10101413</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>Aircraft Technical Development Inc. dba AEROCON Engineering Company</td>
<td>10016409</td>
<td>REV. 5</td>
<td>FAA STC ST0212LA</td>
<td>SEE FAA AMR A321-212/213-231/232 A321-111/112-131/-112-111 A320-231/220-222/223 A320-111/112-212/214 A319-111/112-130/130-133 A318-111/112-130/131-114 A318-111/112-120/122</td>
<td>Installation of Cabin Interior Noise Reduction Kit Revision 5: Addition of model A319-153N in the applicability list.</td>
<td>Certification Basis (CR) for the original product remains applicable. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/appraisal.</td>
<td>- Master Directive AAM-225-NRA319-1530 Rev. K, dated January 11, 2012 (for Continued Airworthiness MMS-A319-1380 Rev. E, dated January 4, 2011) for later FAA approved revisions. Prior to installation of this design change, i.e. it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/07/2020</td>
<td>Active</td>
<td>10016409</td>
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<tr>
<td>Field Aviation Company Inc.</td>
<td>10016419</td>
<td>REV. 3</td>
<td>TCCA C-LSA06-010/D</td>
<td>DHC-8-311</td>
<td>Installation of a 320 cuft baggage compartment and installation of PBE and O2 bottle into the cabin (Ref. TCCA LSTC C-LSA06-010/D).</td>
<td>Installation of 320 cuft baggage compartment is must be performed in accordance with the following: - Modification Instruction 202610 Revision C. Installation of O2 bottle and PBE into the cabin must be performed in accordance with Modification Instruction 202610 Revision C. (Maintenance must be performed in accordance with the Flight Manual Supplement (AFMS), ref. Rosemount Aerospace Doc. D0402888, Revision B, dated 13th July 2009. Rosemount Aerospace EAPAS An...</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/06/2017</td>
<td>Active</td>
<td>10016419</td>
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<td>Bombardier Inc.</td>
<td>10016423</td>
<td>REV. 42</td>
<td>TCCA SA04-112</td>
<td>BD-100-1A10</td>
<td>Installation of Complete Customized Aircraft Interior</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (a) TCCA SCA 2020-04 Dated 8th October 2020 as issued with a CofA dated July 13, 2020.</td>
<td>According to Bombardier Consulting Services Ltd. Certification Definition NoRef D921000 Rev A dated 10 Nov 2019 to be operated in accordance with Aerial Consulting Services Ltd - AFMS D921000 Rev F and/or to be maintained in accordance with the instructions for Continued Airworthiness as referenced...</td>
<td>See those of TCCA STC SA04-112 The FAA approved configurations are identified in the Configuration Definition List document Ref D921000/Aircraft issued with a CofA or on or after 18 May 2019 must be equipped with Helix Free Extinguishers/Prior to installation of this change/repair it must be determined th...</td>
<td>10/10/2022</td>
<td>Active</td>
<td>10016423</td>
</tr>
<tr>
<td>Field Aviation Company Inc.</td>
<td>10016430</td>
<td>REV. 1</td>
<td>TCCA STC C-LSA08-056/D</td>
<td>DHC-8-202, DHC-8-311</td>
<td>Honeywell AHRS to SFIM APIRS Conversion Provision/TCCA STC C-LSA08-056/D, Issue 2</td>
<td>FAAR 25 as per EASA/IMA-191, Issue 5 (inc EWS/EHST Certification Basis for the original product remains applicable to this certificate/appraisal. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain...</td>
<td>Installation of the Conversion Provision for Honeywell AHRS to SFIM APIRS Retrofit must be in accordance with Transport Canada Civil Aviation (TCCA) approved Field Aviation Companies (FACI) Modification/Instructions 21570, Revision B, dated Nov 7, 2011. Conversion from SFIM APIRS back to th...</td>
<td>Installation is limited to Serial Numbers 435, 440, 441 and 407</td>
<td>23/05/2012</td>
<td>Active</td>
<td>10016430</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Foreign/NAA STC id</td>
<td>Model(s)</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10019433</td>
<td>B/E AEROSPACE, Inc.</td>
<td>FAA STC ST0462AT</td>
<td>767-200/300/400</td>
<td>Install the RAAS software in the Runway Awareness &amp; Advisory System (RAAS) Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>This STC is only valid for software versions -224-224 to -226-226. Prior to installation of this design change, it must be determined that the relationship between this design change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness.</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10019433</td>
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<tr>
<td>10019434</td>
<td>B/E AEROSPACE, Inc.</td>
<td>FAA STC ST0459AT</td>
<td>757-200/300</td>
<td>Installation of the Honeywell Runway Awareness &amp; Advisory System (RAAS) Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>This STC is only valid for software versions -224-224 to -226-226. Prior to installation of this design change, it must be determined that the relationship between this design change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness.</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10019434</td>
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<tr>
<td>10019440</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0535AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5177.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915177. Rev. F, dated March 31, 2008 or later FAA approved revision.</td>
<td>Aircraft Serial Number 5177 or later FAA approved Supplement.</td>
<td>06/10/2010</td>
<td>Active</td>
<td>10019440</td>
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<td>10019447</td>
<td>SIERRA INDUSTRIES Ltd.</td>
<td>FAA STC SA8829SW</td>
<td>CESSNA 550 CITATION II</td>
<td>Installation of Interior Rear Assy. on Pilot and Co-Pilot Shoulder Harnesses in accordance with Sierra Master Drawing List SI220-000, Rev. A, dated 28 March 1994. Validation of FAA-approved STC SA8829SW. This is an amendment to EASA.IM.A.S.02793 to include Cessna Model 500 as an approved model.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>06/12/2012</td>
<td>Active</td>
<td>10019447</td>
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<td>10019458</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCAC STC NO. 2A04-113</td>
<td>BD-700-1A10</td>
<td>Customized Interior Completion - ACSN 9272.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>1. TCCA SoC letter C-09-0997 dated 24 November 2009. 2. Reference to the Definitions section of the required AFM Supplement to the referenced FC/0523. 3. WLAN systems are compatible only with IEEE 802.11(b).</td>
<td>18/12/2009</td>
<td>Active</td>
<td>10019458</td>
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<td>10019460</td>
<td>RAUBECK ENGINEERING Inc.</td>
<td>FAA STC SA4362HM</td>
<td>B200CT, B200T, A200C, A200CT, B200, B200C, 200C, 200CT, 200T, A200,</td>
<td>Installation of Butlerfield Industries Oil Residue Collection System KOP/MN B1 101-1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any otherwise installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/08/2018</td>
<td>Active</td>
<td>10019460</td>
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<td>STC Holder</td>
<td>STC Description</td>
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<td>RADBECK ENGINEERING Inc.</td>
<td>Installation of Butterflyfield Industries AirOn Separator System Kit PIN BI 6001-1</td>
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<td>GOODRICH INTERIORS</td>
<td>Installation of Sentry One Flight Deck Door Surveillance Camera System.</td>
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<td>SIERRA INDUSTRIES Ltd.</td>
<td>Plot's Control Wheel Mounted Alternate Gear Horn Silence Switch.In accordance with Sierra Master Drawing List, SI260-000, rev. IR, date 25 Jan 1994 Validation of FAA approval STC SI260-000C, Amend EASA IM AS.02808 to include the Cesna Model 500 as an approved model.</td>
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<td>SIERRA INDUSTRIES Ltd.</td>
<td>Installation of Gas Cylinder Noise bagpase Door Openers in accordance with Sierra Master Drawing List, SI380-00, rev. C, date 24 Sept. 1996. Validation of FAA approval STC SI380-000C, Amend EASA IM AS.02808 to include the Cesna Model 500 as an approved model.</td>
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<tr>
<td>BOMBARDIER AEROSPACE</td>
<td>Observer Seat Installation. Model 409-001 Installation only observer seat between the avionics racks, forward of the main passenger cabin. Revision of EASA STC 10016478. Rev. 1 due to revised technical data (Modification Data Summary F79700 Rev. D).</td>
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<tr>
<td>BOMBARDIER AEROSPACE</td>
<td>Installation of AVS-470 Tail Camera System.</td>
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</table>

**Certification Basis:**
- For Configuration 2 -3 and 5, the modification defined by STC SA43862NM will also require installation of an oil collection tank. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no ad...
<table>
<thead>
<tr>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA</th>
<th>Model/Part No.</th>
<th>Description</th>
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<tr>
<td>BOMBARDIER AEROSPACE</td>
<td>REV.4</td>
<td>TCCA 5A4-113</td>
<td>8D-700-1A10</td>
<td>Aircraft Completion, 13 Passenger Interior Configuration</td>
</tr>
<tr>
<td>BEECHCRAFT CORPORATION</td>
<td>REV. 5</td>
<td>FAA STC 5A10475C</td>
<td>2, C90, E90 1, C90A, C90GT, C90GTI</td>
<td>Increase Operating Gross Weight Approval (as defined in the attached table)</td>
</tr>
<tr>
<td>GOODROCH INTERIORS</td>
<td>REV. 1</td>
<td>FAA STC 5A10475C</td>
<td>737-800</td>
<td>Installation of Hollingswad Santry One Cabin Video Camera System.</td>
</tr>
<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>FAA 5A1434W1-D</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Modification of Collins GPS-4000 or GPS-4000A Global Positioning Sensor (GPS) system installation.</td>
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<tr>
<td>EASA STC No.</td>
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<td>Certification basis:</td>
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<td>BEECHCRAFT CORPORATION</td>
<td></td>
<td></td>
<td></td>
<td>- Increase in gross weight installation in accordance with Cen Tex Aerospace Incorporated (Inc.) Drawing No 030-000-0000, Rev G. Operation in accordance with Cen Tex Aerospace Incorporated (Inc.) Model Flight Manual Supplement defined in Drawing No 030-000-1000, Rev G, For C90A (P54A-190), C90GT, ...</td>
</tr>
<tr>
<td>GOODROCH INTERIORS</td>
<td></td>
<td></td>
<td></td>
<td>- Hollingswad International Drawing List 2712-02, Revision A dated 03/05/2003, Flight Manual Supplement Hollingswad International Inc, document No. 2712-1 AFM, revision N/C dated 03/05/2003, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 260/2004/CF (or...</td>
</tr>
<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td></td>
<td></td>
<td></td>
<td>- Collins Drawing Index RCA-0124-001 Rev-A, dated February 14, 2007. or latest revision level of the Collins Drawing Index - APM Supplement ACC-07-009 Rev-A dated April 2nd, 2007 or latest FAA approved revision, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision...</td>
</tr>
<tr>
<td>1. This STC does not approve WAAS LPV 2. Prior FAA approval/Installation of Collins GPS-4000 part number 822-0391-001-0002-00 or GPS-4000A part number 822-1377-001 is required to this STC. Prior to installation of this modification the installer must determine that the interrelationship below...</td>
<td></td>
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</table>

**Limitations/Conditions**
- Restricted to ASN 9274 and 1) Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS 2) WLAN system is compatible only to IEEE 802.11(b).
<table>
<thead>
<tr>
<th>STC No.</th>
<th>STC Holder/Model</th>
<th>FAA STC</th>
<th>Foreign/NAA STC ref</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10016518</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 2 FAA ST01433W-D</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Modification of Rockwell Collins GPS-4000 or GPS-4000A – RockwellCollins GPS-4000S Global Positioning Sensor installationRevision 1 due to additional models, new Remote Monitoring/Messaging capability, and connection to sensor</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- FAA STC ST01433W-D amended January 24th 2012 - FAA Approved Model List (AML), RICA-039, Revision J, November 08, 2012 - Collins Drawing Index, ref. Doc. RICA-020-001, Rev. K/Airplane Flight Manual Supplement, ref. Doc. ACC-07-010, Rev.8 and later revisions of the above listed documents/approvals...</td>
<td>22/02/2019</td>
<td>Active</td>
<td>10116518</td>
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<tr>
<td>10016523</td>
<td>GOODRICH INTERIORS</td>
<td>REV. 1 FAA STC ST01638LA</td>
<td>737-300, 737-400, 737-500</td>
<td>Installation of Hollingsed Sentry One Cabin mounted Video Camera System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Hollingsed International Master Data List 2708-01, Revision NC, dated 19/05/2004 - Flight Manual Supplement Hollingsed International Inc., document No.2797-AFM, revision Original dated 02/06/2004 or later revisions of the above listed documents/approved by EASA in accordance with EASA ED Decision 2004...</td>
<td>None</td>
<td>07/11/2012</td>
<td>Active</td>
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<tr>
<td>10016535</td>
<td>BRUCE AEROSPACE Inc.</td>
<td>REV. 1 FAA STC ST00941LA</td>
<td>BD-700-1A0</td>
<td>Installation of Floor Proximity Emergency Escape Path Marking System.</td>
<td>EASA IM.A.S.02929 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>EASA STC No. EASA IM.A.S.02912 dated 15 September 2008, FAA STC No. ST00941LA dated 07 May 2008, Bruce Aerospace Master Drawing List No. 710-002 at the following revisions: Rev 0, item 1, dated 1 May 2000, Rev L, item 4, dated 7 July 2000, Rev AN, item 22, dated 14 March 2001, Rev B, item 33, dated 21 Nov...</td>
<td>30/06/2011</td>
<td>Active</td>
<td>10116535</td>
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<tr>
<td>10016544</td>
<td>BOMBARDIER Inc.</td>
<td>REV. 5 TCCA SA08-48</td>
<td>BD-700-1A0, BD-700-1A11</td>
<td>Installation, Interior Completions In-Service Improvement Modifications</td>
<td>EASA IM.A.S.02929 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>TCCA SoC ref. C-11-0232 dated 16 April 2011, TCCA STC No. SA08-48 Issue 2 dated 27 August 2008, AC-61.21, Revision 13, approved revision or later revisions of the above listed documents/approved by EASA in accordance with EASA ED Decision 2004...</td>
<td>None</td>
<td>11/06/2011</td>
<td>Active</td>
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<tr>
<td>10016548</td>
<td>JETAIR AEROSPACE, LLC</td>
<td>REV. 2 FAA STC ST01466LA-D</td>
<td>B737-300-400</td>
<td>Installation of fire protection systems, lower cargo compartment Class D fire protection Class C conversion Original FAA STC ST01466LA-D for B737-300 this revision is implementing the validation of FAA STC ST01466LA-D with B737-300.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>B737-300 - Hollingsed International Master Data List No. ST006-01, Revision NC, dated 26 Jan 2004 - Hollingsed International FAA Approved Flight Manual Supplement No. 2708-AMF, Revision NC, dated 26 Jan 2004, Hollingsed International Instructions for Continued Airworthiness No.2108-14, Revision C, ...</td>
<td>None</td>
<td>29/06/2012</td>
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<td>STC Holder</td>
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<td>System Record</td>
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<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA STC SA7066SW</td>
<td>425 (CONQUEST)</td>
<td>FAA STC SA7066SW</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certification/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions level.</td>
<td>19/11/2014</td>
<td>Active</td>
<td>1011553</td>
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<tr>
<td>BOMBARDIER ING</td>
<td>FAA STC SA5786SW</td>
<td>10016576</td>
<td>FAA STC SA5786SW</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>18/12/2013</td>
<td>Active</td>
<td>1011555</td>
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<tr>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SA1266SE</td>
<td>10016576</td>
<td>FAA STC SA1266SE</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>10/02/2010</td>
<td>Active</td>
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<tr>
<td>LIFEPORT, LLC</td>
<td>FAA STC SA05238E</td>
<td>10016578</td>
<td>FAA STC SA05238E</td>
<td>If more than five vortex generators are damaged or missing, the Aircrafts not airworthy.</td>
<td>30/04/2010</td>
<td>Active</td>
<td>1011678</td>
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<tr>
<td>RAABECK ENGINEERING Inc.</td>
<td>FAA STC SA3875NM</td>
<td>10016580</td>
<td>FAA STC SA3875NM</td>
<td>Prior to installation of this modification the installer must determine the interface relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/02/2019</td>
<td>Active</td>
<td>1011680</td>
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**List of EASA Supplemental Type Certificates**

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<tr>
<th>STC Holder</th>
<th>STC Title</th>
<th>Model</th>
<th>Certification basis</th>
<th>Description</th>
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<tr>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA STC SA7066SW</td>
<td>425 (CONQUEST)</td>
<td>FAA STC SA7066SW</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certification/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions level.</td>
<td>19/11/2014</td>
<td>Active</td>
<td>1011553</td>
</tr>
<tr>
<td>BOMBARDIER ING</td>
<td>FAA STC SA5786SW</td>
<td>10016576</td>
<td>FAA STC SA5786SW</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>18/12/2013</td>
<td>Active</td>
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<tr>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SA1266SE</td>
<td>10016576</td>
<td>FAA STC SA1266SE</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>10/02/2010</td>
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<tr>
<td>LIFEPORT, LLC</td>
<td>FAA STC SA05238E</td>
<td>10016578</td>
<td>FAA STC SA05238E</td>
<td>If more than five vortex generators are damaged or missing, the Aircrafts not airworthy.</td>
<td>30/04/2010</td>
<td>Active</td>
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</tr>
<tr>
<td>RAABECK ENGINEERING Inc.</td>
<td>FAA STC SA3875NM</td>
<td>10016580</td>
<td>FAA STC SA3875NM</td>
<td>Prior to installation of this modification the installer must determine the interface relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/02/2019</td>
<td>Active</td>
<td>1011680</td>
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<tr>
<th>STC No.</th>
<th>STC Holder</th>
<th>FAA STC/ ST no.</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>10016582</td>
<td>RAISSBECK ENGINEERING inc.</td>
<td>FAA STC SA4010NM</td>
<td>REV. 2</td>
<td>2</td>
<td>C90A, C90GT, C90GTi</td>
<td>1, 65-90, 65-A90, 890, E90</td>
<td>Installation of dual aft body strakes on the 90 series KingAir aircraft Raisbeck Engineering inc. STC SA4010 NM.</td>
<td>none</td>
<td></td>
<td>26/04/2011</td>
<td>Active</td>
<td>10016582</td>
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<tr>
<td>10016590</td>
<td>AIRWOLF FILTER CORP</td>
<td>FAA STC SA0258CH</td>
<td>REV. 1</td>
<td></td>
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<td></td>
<td>Installation of Airwolf Filter Corp Air Oil Separator</td>
<td>Installation in accordance with Airwolf Filter Corp.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and models...</td>
<td>26/06/2020</td>
<td>Active</td>
<td>10016590</td>
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<tr>
<td>10016598</td>
<td>BEECHCRAFT CORPORATION</td>
<td>FAA STC ST10596SC-D</td>
<td>REV. 4</td>
<td></td>
<td></td>
<td></td>
<td>Installation of an AirCell Axxess Voice/Data Communications System.</td>
<td>As defined on FAA STC ST10596SC-DF/FAA CAR 49b &amp; 14 CFR Part 25 per EASA TCDS IM.A.065</td>
<td>This modification is only applicable to aircraft equipped with the Collins Proline 21 or Honeywell ST2-800 avionics packages RDL 230-2001-103 and/or 107. Installation requires the aircraft be equipped with Collins Proline 21 radars with the IFIS, STC STO1342-WK-D, and the Data Link Communications option AOA...</td>
<td>30/10/2012</td>
<td>Active</td>
<td>10016598</td>
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<tr>
<td>10019610</td>
<td>BE/AEROSPACE inc. - FSI</td>
<td>FAA ST00004SE</td>
<td>REV. 2</td>
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<td>Remove EWIS limitation for EASA STC No. EASA IM.A.S.02030 - Door 3 Overhead Attendant Rest Installa</td>
<td>The Certification Basis for the original product remains applicable to all aircraft types and model...</td>
<td>The Main deck seating arrangement must be installed in accordance with British Airways drawing 1-302-342 revision 10 dated 12 February 2009 and/or revision 15 dated 12 February 2009.</td>
<td>07/10/2010</td>
<td>Active</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
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<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>Date of Issue</td>
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<tr>
<td>10016625</td>
<td>BEECHCRAFT CORPORATION</td>
<td>REV. 2 FAA SA03622AT</td>
<td>B300, B300C</td>
<td>Installation of an Air Conditioning Evaporator Blower Motor/RF Noise/Filter into a Hawker Beechcraft Model B300 / B300C aircraft</td>
<td></td>
<td></td>
<td>In accordance with Master Data List Number HBS 063-2151-101-00 Revision 1.0, dated 06/16/08 or later approved revision.</td>
<td>Prior to installation of this modification I must be determined that the relationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>31/10/2014</td>
<td>Active</td>
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<tr>
<td>10016626</td>
<td>TRUATLANTIC MFG., LLC</td>
<td>REV. 2 FAA ST00026AT</td>
<td>208, 208B</td>
<td>Installation of Passenger Interior and Cargo Door Modification in Cessna Caravan (Cargo Conversion)</td>
<td></td>
<td></td>
<td>Installation in accordance with Atlantic Aero Master Drawing List208199203, Revision B, dated October 28, 1993 FAA approved Aircraft Flight Manual Supplements a) 20893057, Rev. A, dated December 13, 1993 (Cessna 208, S/N 20880016 and on) b) 20893058, Rev. A, dated December 13, 1993 (Cessna 208, S/N 2.) a) This approval should not be extended to other aircraft of this model which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse effect...</td>
<td>28/04/2016</td>
<td>Active</td>
<td>10016626</td>
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<tr>
<td>10016627</td>
<td>TRUATLANTIC MFG., LLC</td>
<td>REV. 2 FAA STC SA00154AAT</td>
<td>208, 208B</td>
<td>Installation of Cabin Windows in the Cargo versions of Cessna 208 series.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Atlantic Aero Master Drawing List. Rev. G, dated November 16, 2008, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Manual...</td>
<td>a) This approval should not be extended to other aircraft of this model which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse effect...</td>
<td>28/04/2016</td>
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<tr>
<td>10016646</td>
<td>ROCKWELL COLLINS AEROSPACEAND ELECTRONICS, INC.</td>
<td>REV. 2 FAA STC ST01100SE</td>
<td>DHC-8-401 &amp; DHC-8-402</td>
<td>Updates for Flight Management System (FMS) HGS 4100: FAA acceptance letter 1306-00-387, dated 21.09.2009.</td>
<td></td>
<td></td>
<td>EASA approved AFM Supplement No. 9701-1192, Rev. C or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>None</td>
<td>12/04/2010</td>
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<td>10016646</td>
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<tr>
<td>10016648</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>REV. 2 TCCA C-LISA08-0180</td>
<td>DHC-8-402</td>
<td>Galley (G3) Water System Installation</td>
<td></td>
<td></td>
<td>Installation or Modification of the Galley G3 Water System must be in accordance with TCCA approved FACI Master Data List C20300, Revision B, and/or for Continued Airworthiness Maintenance must be in accordance with FACI C2A 21303, Revision B, and/or for continued airworthiness maintenance...</td>
<td>Limited to Ser. Nr. 4054, 4062, 4063, 4183, 4224, 4236, 4273, 4286, 4290, 4337, 4392 Prior to installation of this design change it must be determined that whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect up...</td>
<td>28/03/2014</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10016649</td>
<td>DECRANE AEROSPACE PATS AIRCRAFT, LLC</td>
<td>REV. 1</td>
<td>FAA STD STO2057LA</td>
<td>BOEING 737-800 SERIES BOEING 737-700 SERIES</td>
<td>Installation of Electronic Equipment Rack in All Bulk Cargo Compartment This STC revises and supercedes EASA STC EASA.IM.A.S.03093. The revision was issued to reflect the transfer from Hollingshead International to Decrane Aerospace PATS Aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Hollingshead Master Data List 2876-01 Rev. A dated June 07, 2007 Instructions for Continued Amnorphness 2876-14 Rev. A dated June 14, 2007 for later revision of the above document approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>None.</td>
<td>18/11/2009</td>
<td>Active</td>
<td>10116649</td>
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<tr>
<td>10016651</td>
<td>LEARJET Inc.</td>
<td>REV. 2</td>
<td>FAA STC SA8199NM-D</td>
<td>BD-700-1A10</td>
<td>FAA STC SA8199NM-D. Installation of a Full Authority Cabin Temp Control. EASA TCDS EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. SA8199NM-D dated 01 May 2002 – FAA SoC reference SU-08-394 dated 10 December 2008, Bombardier Master Document List No. BAS63010007, Revision D, dated 24 August 2004 or later approved revision - Bombardier AFMS No. BAS67040007, Revision NC or later approved revision or later revisions of this document.</td>
<td>None.</td>
<td>16/12/2010</td>
<td>Active</td>
<td>10116651</td>
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<td>10016654</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC STD1276W-D</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Modification of Rockwell Collins Communication Management Unit (CMU-500) Aircraft Communications Addressing and Reporting System (ACARS) Core Software Rev. 1 updates the CMU-500 from Core 14 to Core 16</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Based on 14 CFR §§ 21.115 and 21.101, and the FAA policy for significant changes in FAA Order 8110.48, the certification basis for the Pilatus PC-12 passenger to Freighter is as follows: a. The type certification basis for the Pilatus PC-12 airplane is shown on TCDS A/9EU for parts not changed or not affected.</td>
<td>1. Drawing Index ref. RIC.0445-001 revision F2, Approved Model List RICA-0445 revision G.</td>
<td>19/05/2020</td>
<td>Active</td>
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<td>10016658</td>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC STO3980AT</td>
<td>PC-12</td>
<td>Installation of a Flight Components/EMTEQ LED Position (Navigation) lights into a Pilatus PC-12 Series Aircraft and associated hardware. Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>Based on 14 CFR §§ 21.115 and 21.101, and the FAA policy for significant changes in FAA Order 8110.48, the certification basis for the Pilatus PC-12 passenger to Freighter is as follows: a. The type certification basis for the Pilatus PC-12 airplane is shown on TCDS A/9EU for parts not changed or not affected.</td>
<td>1. Master Drawing List 2-59640-001 Revision A dated October 13, 2008 or later FAA approved drawing revisions. 2. Instructions for Continued Amnorphness (ICA) EMTEQ Engineering Document No. 2-59640-905. Rev. III dated July 11, 2008 or later FAA accepted revision.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the amnorphness of the product.</td>
<td>02/03/2016</td>
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<td>EASA STC no.</td>
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<td>Holders</td>
<td>Rev.</td>
<td>Models</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10016660</td>
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<td>BOMBARDIER INC.</td>
<td>REV. 1</td>
<td>TCCA STC SA05-40</td>
<td>Installation, Complete Custom Aircraft Interior (S/N 9311).</td>
<td>EASA STC No. EASA.IM.A.S.03121.TCCA STC No. SA05-60 Issue 1.Aero Consulting Services Ltd Configuration Definition List No. E025000.Revision CT dated 31 May 2010.or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of...</td>
<td>Restricted to ASN 9311.</td>
<td>04/06/2010</td>
<td>Active</td>
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<td>10016665</td>
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<td>B/E AEROSPACE LIMITED</td>
<td>REV. 1</td>
<td>TCCA C-LSA09-031/D, ISS, 2</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>TCCA LSTC No. C-LSA09-031/D, Issue 2 dated 17 June 2013.Aero Consulting Services Ltd Modification Data Summary 000000 Revision 1 Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, 33-299M007 Rev NC.Aero Consulting Services Ltd.</td>
<td>Restricted to ASN 8088.Prior to installation of this modification the installer must determinethat the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/08/2013</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAACertification basis</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>ACS-NAI LLC</td>
<td>REV. 1</td>
<td>FAA STC LSA09-0050 ISSUE 2</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>FAA STC LSA09-0050 Issue 2 dated 28 September 2013</td>
<td>Restricted to ASN 8081. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>24/05/2013</td>
<td>Active</td>
<td>10016679</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST03646AT-D</td>
<td>Installation of Cabin/Baggage Compartment</td>
<td>FAA STC ST03646AT-D</td>
<td></td>
<td>08/02/2012</td>
<td>Active</td>
<td>10016679</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST03646AT-D</td>
<td>Installation of Cabin/Baggage Compartment</td>
<td>FAA STC ST03646AT-D</td>
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<td>GLOBAL AIRWORKS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST01585LA</td>
<td>Installation of AirWorks Cockpit Door Surveillance Systems (Validated to the FAA STC ST01585LA). This STC installs a CDS.</td>
<td>FAA STC ST01585LA</td>
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<td>30/11/2011</td>
<td>Active</td>
<td>10016884</td>
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<td>FLY IN VACATIONS</td>
<td>REV. 1</td>
<td>FAA STC SED031NM</td>
<td>Installation of turbocharger and induction air intercooler in accordance with &quot;Drawing List Report No. 1&quot; Dated August 20, 1990, or later FAA approved revision.</td>
<td>FAA STC SED031NM</td>
<td></td>
<td>08/06/2015</td>
<td>Active</td>
<td>10016700</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>FAA STC NO</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10016706</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>REV. 1</td>
<td>FAA STC NO SR01424SE</td>
<td>Cargo Hook Kit models 200-292-01</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificates of noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>155-156/00, revision 11 dated 9 April 2010 - Master Drawing List 130-112-01 revision 4 dated 28 May 2010 - Owners Manual; 133-017-01 revision 4 dated 28 May 2010 - Specifications for Continued Airworthiness; 122-050-00 revision 17 dated 9 March 2010 - Cargo Hook Service Manual 121-019-01 revision 1 dated...</td>
<td>None</td>
<td>21/02/2010</td>
<td>Active</td>
<td>1017670</td>
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<tr>
<td>10016707</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>REV. 2</td>
<td>FAA STC NO SR00696SE</td>
<td>Fabrication of Onboard Systems Model 200-246-00 cargo hook kit.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SR00696SE. The requirements for environmental protection and the associated certificates of noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Onboard Systems Master Drawing List No. 155-047-00, Revision 21, dated 21 March 2013 (&gt;): &gt; or later FAA approved revision Onboard Systems Service Manual No. 120-083-00, Revision 9, dated 15 February 2013, or later FAA approved revision Section 4 Onboard Systems Service Manual No. 122-004-00, Revision 21 date...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between the design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The STC is valid for those aircraft types and mo...</td>
<td>21/06/2019</td>
<td>Active</td>
<td>1017670</td>
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<tr>
<td>10016714</td>
<td>DONALDSON COMPANY, Inc.</td>
<td>REV. 2</td>
<td>FAA STC NO SR09416HC</td>
<td>Installation of Aerospace Filtration System (AFS) Engine Inlet Barrier Filter (IBF) system. Revision 2 reflects the changes and address changes from Aerospace Filtration Systems, Inc. to Donaldson Company, Inc.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and associated certificates of noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>1. AFS Drawing Number 19860, Revision B dated 17/6/2004 or later approved revision 2. Installation Procedure AFS-BH006L34-4IBF-KIT-IP, initial released dated 16/8/2004 or later approved3. Rotorcraft Flight Manual Supplement (RFMS) AFS-BH006L34-4-IBF-KIT-FMS Initial release dated 25/8/2004 or later appro...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between the design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>3/01/2018</td>
<td>Active</td>
<td>1017674</td>
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<td>10016715</td>
<td>DONALDSON COMPANY, Inc.</td>
<td>REV. 2</td>
<td>FAA STC NO SR09459RC</td>
<td>Installation of Aerospace Filtration System (AFS) Engine Inlet Barrier Filter (IBF) system. Revision 1 reflects the changes and address changes from Aerospace Filtration Systems, Inc. to Donaldson Company, Inc. Revision 2 updates the Associated Technical Documentation with regard to “later revisions”</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and associated certificates of noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>1. AFS Drawing Number 198600, Revision A dated 8/4/2003, 2 Installation Procedure AFS-BH047-IBF-KIT-IP, initial released dated 11/4/20033. Rotorcraft Flight Manual Supplement (RFMS) AFS-BH407-IBF-KIT-FMS Initial release dated 15/12/20034. Instructions for continued airworthiness (ICA) AFS-BH407-IBF-K...</td>
<td>Rotorcraft Flight Manual Supplement dated 10/12/2003 is a required part of this STC. “Instructions for continued airworthiness” AFS-BH407-IBF-KIT-ICA is an approved part of this STC. Prior to installation of this design change, it must be determined that the interrelationship between the design change and...</td>
<td>10/11/2021</td>
<td>Active</td>
<td>1017675</td>
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<tr>
<td>10016725</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>REV. 1</td>
<td>TCCA STC SH 96-32</td>
<td>AS350 BA, AS350 B2, AS350 B3</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificates of noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Configuration and Installation Documentation - Eurocopter Canada Limited Master Drawing List No. MDA-063-03, Rev. UMaintenance documentation - Eurocopter Canada Limited Information Continuous Airworthiness No ICA-ECL-1168 Rev. 2Operating documentation - AS350BA any serial number and AS350B2 pre-mod...</td>
<td>None</td>
<td>13/12/2010</td>
<td>Active</td>
<td>1017675</td>
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<td>STC Holder</td>
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<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>OMOBADO SYSTEMS INTERNATIONAL</td>
<td>EC130B4, AS350B1, AS350B, AS350B2, AS350B3, AS350B5</td>
<td>Installation of Cargo Hook Kit Model 200-261-00, 200-261-01 &amp; 200-353-0076. This STC is a revision to EASA STC EASA.IM.R.S.01027 Revision 1 through the validation of amendment dated 11 August, 2009 to FAA STC SR008865E. The original product certification basis as per EASA TCDS R.038 dated 10 November, 2008. The Certification Basis for the original product remains applicable to this certificate/ approval. The modified noise and/or emissions levels of the product unchangeable remain applicable to this certificate/ approval.</td>
<td>FAA STC SR008865E</td>
<td>FAA STC SR008865E</td>
<td>-</td>
<td>10/08/2023</td>
<td>Active</td>
<td>10101729</td>
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<tr>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>AS350B1, AS350B2, AS350B3, AS350B4, AS350B5</td>
<td>Installation of LH and/or RH Cargo Pods Rev. 1: Minor Changes embodied and new name of STC (Rev. 2: added Compatibility to sliding door with the LH RH Cargo Pod installation for AS350 B2 and AS350 B3 helicopters only. The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product unchangeable remain applicable to this certificate/ approval.</td>
<td>FAA STC SR01928CH</td>
<td>TCCA SH97-15</td>
<td>-</td>
<td>10/06/2020</td>
<td>Active</td>
<td>101016731</td>
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<tr>
<td>FARGO MANUFACTURING CO.</td>
<td>MD900</td>
<td>Install 61.8 gallon crashworthy auxiliary fuel tank on the aft cabin baggage bay floor. The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product unchangeable remain applicable to this certificate/ approval without any impact.</td>
<td>FAA STC SR01928CH</td>
<td>MD900</td>
<td>-</td>
<td>10/06/2020</td>
<td>Active</td>
<td>101016731</td>
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<td>SAFRAN ELECTRONICS &amp; DEFENSE AVIONICS USA, LLC</td>
<td>EC120B</td>
<td>SFM two-axis VFR Autopilot System (SAGEM PA85)(Note: - Original STC FAA STC SR00630RC-D, amended 9 February, 2011. Existing EASA STC: 10016732 rev 3). Revision 4 incorporates 2 new configurations related to minor changes to the FAA STC. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product unchangeable remain applicable to this certificate/ approval.</td>
<td>FAA STC SR00630RC-D</td>
<td>EC120B</td>
<td>-</td>
<td>14/03/2011</td>
<td>Active</td>
<td>101016732</td>
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<td>DONALDSON COMPANY, Inc.</td>
<td>369MH, 369H, 369D, 369E, 369FF, 369H, 369HE</td>
<td>Installation of Aerospace Filtration Systems Inlet Barrier Filter System ahead of Engine Air Separator (with or without Mid Eliminator) or PDScreen Revision 1 reflects the name/address change from Aerospace Filtration Systems, Inc. to Donaldson Company, Inc. The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product unchangeable remain applicable to this certificate/ approval.</td>
<td>FAA STC SR01364A</td>
<td>369MH, 369H, 369D, 369E, 369FF, 369H, 369HE</td>
<td>-</td>
<td>31/01/2018</td>
<td>Active</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Rev. level</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EUROCOPTER</td>
<td>A109S</td>
<td>Rev. 1</td>
<td>Installation of alternative Swivelling EMS Equipment P/N 145146-501 for EMS Equipment P/N 007-002098-501 for EMS Equipment P/</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Flight Manual Supplement - EMS Equipment P/N 145146-501, Revised 1</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the acceptability of the product. The installation of this modification by third party.</td>
<td>20/10/2013</td>
<td>Active</td>
<td>10016741</td>
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<td>AEROLITE AG</td>
<td>AS355 F2</td>
<td>Rev. 2</td>
<td>Installation of alternative Swivelling Floor Plate Assy P/N 007-002088-501 for EMS Equipment P/N N45181-501</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Flight Manual Supplement - EMS Equipment P/N 145146-501, Revised 1</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the acceptability of the product. The installation of this modification by third party.</td>
<td>20/10/2013</td>
<td>Active</td>
<td>10016750</td>
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<td>AEROLITE AG</td>
<td>A109S</td>
<td>Rev. 1</td>
<td>Installation of alternative Swivelling EMS Equipment P/N 145146-501 for EMS Equipment P/N 007-002098-501 for EMS Equipment P/</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Flight Manual Supplement - EMS Equipment P/N 145146-501, Revised 1</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the acceptability of the product. The installation of this modification by third party.</td>
<td>20/10/2013</td>
<td>Active</td>
<td>10016764</td>
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<td>AEROLITE AG</td>
<td>AS355 E/F/I/F/I/F2</td>
<td>Rev. 3</td>
<td>Installation of alternative Swivelling Floor Plate Assy P/N 007-002088-501 for EMS Equipment P/N N45181-501</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Flight Manual Supplement - EMS Equipment P/N 145146-501, Revised 1</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the acceptability of the product. The installation of this modification by third party.</td>
<td>20/10/2013</td>
<td>Active</td>
<td>10016765</td>
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<tr>
<td>AEROLITE AG</td>
<td>AS355 F</td>
<td>Rev. 1</td>
<td>Installation of alternative Swivelling Floor Plate Assy P/N 007-002088-501 for EMS Equipment P/N N45181-501</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Flight Manual Supplement - EMS Equipment P/N 145146-501, Revised 1</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the acceptability of the product. The installation of this modification by third party.</td>
<td>20/10/2013</td>
<td>Active</td>
<td>10016770</td>
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<td>AEROLITE AG</td>
<td>AS355 F</td>
<td>Rev. 1</td>
<td>Installation of alternative Swivelling Floor Plate Assy P/N 007-002088-501 for EMS Equipment P/N N45181-501</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Flight Manual Supplement - EMS Equipment P/N 145146-501, Revised 1</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the acceptability of the product. The installation of this modification by third party.</td>
<td>20/10/2013</td>
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<td>10016774</td>
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<td>EASA STC no</td>
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<td>Rev.</td>
<td>Foreign/NA A</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10016796</td>
<td>HELI-ONE (NORWAY) AS</td>
<td>REV. 1</td>
<td>AS 332 L, AS 332 L1</td>
<td>BPOL 332/1 Police Equipment Installation as per MOD. AHSE-S2408Rev. 0 of this STC consists of the full installation of the equipment listed in Configuration Control AHSE-S2408Rev. 1 of this STC, in accordance with AR2E-69487, allows installation of the following equipment only: AF-155 Autopilot instal...</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA approved Airfilm Single Pole Utility Bracket Install Revision 1 of this STC consists of the full installation of the equipment listed in Configuration Control AHSE-S2408 Rev. 1 of this STC, in accordance with AR2E-69487, allows installation of the following equipment only: AF-155 Autopilot instal...</td>
<td>None</td>
<td>24/11/2015</td>
<td>Active</td>
<td>10101796</td>
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<td>10016798</td>
<td>SKYTRAC SYSTEMS LTD.</td>
<td>REV. 4</td>
<td>TCCA STC SHK3-30</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of SkyTrac ISAT-100/ ISAT-200 AIRBORNE POSITION/ DATA COMMUNICATOR Revision 1 to add: BK-17 A-1, B-1, and B-2 models</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition on Lithium Batteries as per EASA CJ-01 issue 2The requirements for environmental protection and the associated certification noise and/or emissions levels.</td>
<td>FAA approved EASA STC 10016801</td>
<td>06/04/2018</td>
<td>Active</td>
<td>10101798</td>
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<td>10016801</td>
<td>AIRFILM CAMERA SYSTEMS, INC</td>
<td>REV. 1</td>
<td>FAA STC SRO1785LA</td>
<td>SEE FAA ANL</td>
<td>Installation of Airfilm Single Pole Utility Bracket Revision 1 of this STC (former EASA STC SRO1785LA) is issued to validate FAA STC SRO1785LA amended to incorporate minor changes to the type design data.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA approved Master Drawing List No. AFSP-010, Revision N/C, dated 17 February 2005 and a copy of FAA approved Rotorcraft Flight Manual Supplement No. AFSP-005, issued 12 May 2005 or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherpreviously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/06/2018</td>
<td>Active</td>
<td>10101801</td>
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<td>10016802</td>
<td>AIRFILM CAMERA SYSTEMS, INC</td>
<td>REV. 1</td>
<td>FAA STC SRO1654LA</td>
<td>407 206, 206L-1, 206L-3, 206L-4 206A, 206B, 206L-1, 206L-3, 206L-4</td>
<td>Installation of Airfilm Model G1 Nose Mount Utility Bracket for G3 Landing Light on Bmd 206/407 Series Rotorcraft</td>
<td>Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC.</td>
<td>FAA approved Airfilm Camera Systems Master Drawing List No. AFSP-005, issued 12 May 2005 or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/10/2018</td>
<td>Active</td>
<td>10101802</td>
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<td>10016805</td>
<td>GENESYS AEROSYSTEMS, INC.</td>
<td>REV. 2</td>
<td>FAA SRO2209AK</td>
<td>BELL 407 BELL 206L-4 BELL 206L-3 BELL 206L-1 BELL 206L BELL 206B BELL 206A 206-2, 206A, 205, 205A, 206L-1</td>
<td>Installation of Genesys Aerosystems FlightLogic Electronic Flight Information System (EFIS) Rev. 2 upgrades the software to version 9.0A on the dual Integrated Display Unit (IDU)-450 and IDU-601 configurations.</td>
<td>Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Data List ref. 651-04556 Rev. 2, dated 10 April 2018- Flight Manual Supplements. Document 17-209M5-0450 (IDU-450). Rev. A, dated 12 July 2021 - Document 17-209M5-0690 (IDU-601). Rev. A, dated 12 July 2021</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/11/2021</td>
<td>Active</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<td>AEROLITE AG</td>
<td>REV. 5</td>
<td>MBB-BK117 C-2 MB-BK 117 C-2</td>
<td>EMS Equipment P/N 145212-501 Installation of EMS Equipment P/N 145212-501 with various configurations. Rev. 2 introduced the following changes to the EMS Equipment P/N 145212-501 (EASA approved under STC No. 1001617, Rev. 1): (1) Modification of configuration no. 2. (2) Introduction of derivative config...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Findings: (a) CR D-01 (Passenger Emergency Exit - EMS Installation) is raised and closed at issue 3. The requirements for environmental prot...</td>
<td>Certification Program CPR-14-007 Rev. N/C dated 21/07/2014(+) - Certification Program CPR-21-038 Rev. E dated 16/09/2022(+) - Flight Manual Supplement FMS-06-002 Rev. F dated 01/08/2022(+) - Instructions for Continued Airworthiness and Operations ManualPA-06-002 Chapter 04 dated 13/04/2006(+)</td>
<td>The following are pre-requisite installations for the embodiment of the design change leading to the Rev. 5 of this STC 1001617: Airbus Helicopters Deutschland (AHD) floor (*) for the fireprotection installation P/N 145-0047223-504 - Cabinet provisions P/N 145-0052560-501 for the cabinet instalat...</td>
<td>20/08/2022</td>
<td>Active</td>
<td>1016737</td>
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<td>DART AEROSPACE Ltd.</td>
<td>REV. 1</td>
<td>TCCA SH03-6, ISSUE 4</td>
<td>3. 214B, 214B-1 2. 212, 412, 412EP 1. 205B, 205A-1 D205-594 Extended height landing Gear/Pod: This certificate revision of STC EASA.IM.R.S.01198 forextension of applicability to 412 and 412EP helicopter models...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Document List MDL-D205-594 Rev. 2 dated 17 June 2013 - Installation Instructions RN-D205-594 Rev. E dated 7 May 2013-Instructions for Continued Airworthiness ICA-D205-594 Rev. 4 dated 7 May 2013-Flight Manual Supplement FMS- D214-594 Rev. A dated 14 July 2005or later revisions of the ab...</td>
<td>Prior to installation of this design change it must be determined if there is an interrelationship between this design change and any other previously installed design change and/or repair will introduce inadvertent effect upon the airworthiness of the product.</td>
<td>28/01/2014</td>
<td>Active</td>
<td>1018526</td>
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<td>APICAL INDUSTRIES, Inc. doing business DART AEROSPACE</td>
<td>REV. 2</td>
<td>FAA STC SR01855LA EC135 T2+, EC135 T3 EC135 P3, EC135 T1, EC135 T2 EC135 P1, EC135 P2, EC135 P2+ Emergency Floats Kit Installation of Apical Industries, Inc. Emergency Floats Kit in accordance with FAA STC SR01855LA Apical Master Document List MD-135-1provides for different installations either with floats or without. REV. 1: Amendment of the initial STC in order to add the EC135 MidHeight...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Rev. 1 - Rotorcraft Flight Manual Supplement RMS-135-1, Rev. C - Flight Manual Supplement RMS-135-2, Rev. F - Airworthiness Limitations Section 2.0 of ICA-D214 Faber, Rev. O: later revisions of the above listed documents approved by EASA in accordance with ICA-D214-594-Flight Manual Supplement FMS- D214-594, Rev. A dated 14 July 2005or later revisions of the ab...</td>
<td>31/07/2017</td>
<td>Active</td>
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<td>CERT WORKS, LLC</td>
<td>REV. 1</td>
<td>FAA SR016475E EUROCOPTER FRANCE AS 350 B2 Installation of Honeywell LTS-101-7003-2 engine on AS 350 B2 Rev. 1. Transfer of STC to Cert Works, LL...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval without any...</td>
<td>Master Drawing List 960-106, Revision C, dated September 25, 2006Maintenance &amp; Overhaul Supplement S2021H, Revision 1, dated September 2006-Rotorcraft Flight Manual Supplement S2021H, Revision 1, dated April 6,2006or later revisions of the above listed document(s) approved/ accepted on behalf of EASA See attached Annex Issue 01 dated 27/11/2020. Prior to installation of this design change/repair it must be determined if there is an interrelationship between this change/repair and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.</td>
<td>27/11/2020</td>
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<td>1018464</td>
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<td>PRECISE FLIGHT, Inc.</td>
<td>REV. 1</td>
<td>FAA STC SH0139NM</td>
<td>SEE EASA APPROVED MODEL LIST Precise Flight PulseLift Control System. Adding MDH 399 and 500 helicopters...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Multiple (see attached sheet)</td>
<td>Approval for the physical installation of the PulseLift Control Unit will be required if there is no specific reference provided in Engineering Drawing List 0019P001, revision B: dated May 1, 2002, orlater FAA approved revision.</td>
<td>08/04/2011</td>
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<td>EASA STC no.</td>
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<td>Rev.</td>
<td>Foreign/NAA STC no.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>10016700</td>
<td>OMBARO SYSTEMS INTERNATIONAL</td>
<td>REV. 1</td>
<td>FAA STD SR01303SE</td>
<td>AS350B, AS350B1, AS350B2, AS350B3, AS350D</td>
<td>Onboard Systems cargo hook kits P/N 200-286-01 (with cargo hook P/N 200-286-03) and P/N 200-286-02 (with cargo hook P/N 528-029-00) Revision 1 - introduces cargo hook kit P/N 200-286-01 STC.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: Change under revision 1 complies with FAR Part 27, amends 27.1- through 27.397 the requirements for environmental protection and the associated...</td>
<td>None</td>
<td>04/02/2011</td>
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<td>10016782</td>
<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>REV. 3</td>
<td>FAA SR01812LA</td>
<td>A109/ A109A/ A109AII/ A109C</td>
<td>Installation of Apical Industries, Inc. Emergency Float Kit Revision 2 of this STC is issued to add additional models and introduced features change.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. There are requirements for environmental protection and the associated certified noise and/or emissions levels of the original product...</td>
<td>None</td>
<td>09/04/2021</td>
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<td>1011878</td>
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<td>10016879</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>REV. 1</td>
<td>FAA STC SR02408N-D</td>
<td>S-92A</td>
<td>Installation of Spectrolab Nights II XP Searchlight/Note: This certificate revises and supersedes STC No. EASA.IM.R.S.01255, aiming to update the Associated Technical Documentation as listed below.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Master Document List: MD105-1, Revision D, dated 15 August 2005; Rotorcraft Flight Manual Supplement, RFMSA109-1, Revision N/C, dated 24 August 2005; Instruction for Continued Airworthiness, CA109-1, Revision A, dated 19 August 2005; later revisions of the above listed documents approved.</td>
<td>09/04/2021</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>DART AEROSPACE Ltd.</td>
<td>TCA STC: SH06-30</td>
<td>Installation of Multiple-place Utility Troop/Seating (System Rev 1)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCA STC No. SH06-30 Issue 2 dated 2 July 2009FMS-D119-756-EASA Rev. A, ICA-D119-746-EASA Rev. 0, MDL-D119-756-EASAeff.</td>
<td>None</td>
<td>10/02/2011</td>
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<td>BAE SYSTEMS LAND &amp; ARMSMNTS L.P.</td>
<td>FAA SR019016LA</td>
<td>Initial issue) Installation of Multiple-place Utility Troop/Seating Design Rev 2)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Drawing Number TD-05175 Revision P dated September 24, 2009 MB-BK-117 C-2 Rotorcraft Flight Manual Supplement Document No. TD-05174 Revision D dated December 10, 2009 MB-BK-117 D-2 Rotorcraft Flight Manual Supplement Document No.14150-36-2 initial date issued February 3, 2010 or later...</td>
<td>- A copy of this STC must be included in the permanent records of the modified aircraft. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the person written evidence of that permission. The following BAE Systems Aerospace &amp; Defense Grou</td>
<td>07/10/2021</td>
<td>Active</td>
<td>10101617</td>
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<td>DONALDSON COMPANY, Inc.</td>
<td>FAA STC SR049016C</td>
<td>Installation of Aerospace Filtration Systems (AFS) Engine Inlet Barrier Filter (IBF) on Bell 206B helicopters Revision 1 reflects the name and address change from BAE Systems Aerospace &amp; Defense Group Inc. to BAE Systems Land &amp; Armaments L.P.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. AFS drawing number 109900 revision A dated August 16, 2005 or later approved revision 2. Installation procedures AFS-BH08B-IBF-KIT-P, revision A dated October 25, 2005 or later approved revision 3. Instruction for continued airworthiness document AFS-BH08B-IBF-KIT-ICA, are required. 4. Airworth...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>3/10/2018</td>
<td>Active</td>
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<td>ONSBOARD SYSTEMS INTERNATIONAL</td>
<td>FAA STC SR011648E</td>
<td>Installation of External Load Swing Suspension System models 200-280-01 and 04 (with Keepers) Cargo Hooks and model 200-280-02 and -03 (with Hydraulic Cargo Hooks) Revision 1 introduced cargo hook kit 200-280-03 with a collective or friction knob and a new release lever to accommodate the AS350B3 collective/collective con...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Definition and installation Onboard Systems International Master Drawing List No. 155-086-00, Rev. 05, dated 12 May 2021 Onboard Systems International Owner’s Manual No. 120-104-01 Rev. 17, dated 9 May 2014 (for system model 200-280-01) (rev. 120-104-02 Rev. 37, dated 10 May 2021 (for system model 2)...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product installation and operation of the system on AS350B3 vs...</td>
<td>13/06/2022</td>
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<td>10101637</td>
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<td>EAGLE COTHERS USA, INC.</td>
<td>FAA SH47478NM</td>
<td>Installation of sensors, toggle switches, circuit breakers and replacement wiring harnesses in accordance with Genavair Aviation Master Drawing List No. GA107, This STC is the validation of FAA STC No. SH47478NM (issued on July 29, 1989) Revision 2 follows re-issuance of FAA STC No. SH47478NM on October 28, 2018...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Genavair Aviation Master Drawing List (MDL) GA107, Revision Q dated February 12, 2013 (Rev. 3) AS5608, AS5609, AS5600, AS5681, AS5682, AS5683 with P122 Consensus and PREE MOD AMS 07-420 configuration - Rotorcraft Flight Manual Supplement GA157-2 Revision N dated 04/03/2011...</td>
<td>In all cases where works have been carried out on the collective load locking strap or central console- the locking strap height must be adjusted as required in AH Maintenance Manual 67.10.00.501 or 67.10.00.501 (as appropriate) section 3.3, arm- a minimum gap of 3.15 inches (44cm) between the end of...</td>
<td>18/02/2016</td>
<td>Active</td>
<td>10101645</td>
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## List of EASA Supplemental Type Certificates

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<th>STC No.</th>
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<td>10016690</td>
<td>AIR COMM CORPORATION L.L.C</td>
<td>REV. 1</td>
<td>FAA STC SR0066SE</td>
<td>EC135 T2, EC135 T2+ EC135 T2, EC135 T2+ EC135 P2, EC135 P2+</td>
<td>Installation of Cabin Air Conditioning System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SR0066SE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Air Com Corporation Master Drawing List Report No.DL-EC135, Revision A, dated 19 October 2004. FAA approved 24 November 2004, or later FAA-approved revision instructions for Continued Airworthiness, Air Com Corporation ReportEC136-200MM-1, Revision - , dated 25 August 2004. FAA approved 24 November 2000.</td>
<td>23/06/2020</td>
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<td>10016690</td>
<td>EAGLE COPTERS USA, INC.</td>
<td>REV. 1</td>
<td>FAA STC SR00471SE</td>
<td>AS335F2, AS335F1, AS335F</td>
<td>Modular Equipment Rack Installation per MOL GA170 Notes 1. The STC was initially issued to Geneva Aviation, Inc. (initial reference: EASA.IM.R.S.01357)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval . The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Master Drawing List GA170 Rev. B, dated November 20, 2002.FMAC Flight Manual Supplement GA170-3 Rev. D, Dated 02/24/2004. or subsequent revisions of this decision.</td>
<td>This STC provides approval for structural provisions only. A separate approval is required for the installation of any equipment in the modular equipment rack. Prior to installation of this design change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce harmful effect upon the airworthiness of the product.</td>
<td>12/03/2015</td>
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<td>10016691</td>
<td>EAGLE COPTERS USA, INC.</td>
<td>REV. 1</td>
<td>FAA STC SR00326SE</td>
<td>AS335F2, AS335F1, AS335F</td>
<td>Microwave System Installation per MOL GA170 Notes 1. The STC was initially issued to Geneva Aviation, Inc. (initial reference: EASA.IM.R.S.01359)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Master Drawing List GA170 Rev. D, dated February 11, 2009.FMAC Flight Manual Supplement GA170-3 Rev. B, dated May 14, 2010, or subsequent revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/49/EC.</td>
<td>Prior to installation of this design change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce harmful effect upon the airworthiness of the product.</td>
<td>12/03/2015</td>
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<td>10016692</td>
<td>DART AEROSPACE USA, INC.</td>
<td>REV. 1</td>
<td>FAA STC SH4748NM</td>
<td>AS335B, AS335B1, AS335B2, AS335B3, AS335A, AS335A1, AS335B1, AS335B2, AS335B3, AS335A</td>
<td>Installation of a replacement instrument panel visor on the STC holders.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Geneva Aviation Master Drawing List, doc. ref. GA127, Revision D, dated March 8, 2008. FAA approved 17, 2008.</td>
<td>None</td>
<td>15/04/2013</td>
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<td>10016693</td>
<td>EAGLE COPTERS USA, INC.</td>
<td>REV. 1</td>
<td>FAA STC SR00222SE</td>
<td>AS335B, AS335A, AS335B1, AS335B2, AS335B3, AS335B1, AS335B2, AS335B3</td>
<td>Installation of a supplemental avionics console between the forward seat and the console associated with Geneva Aviation Master Drawing List GA156. This STC is the validation of FAA STC No. SR00222SE. Revision 1 follows re-issuance of FAA STC No. SR00222SE on October 28, 2015. It confirms the transfer of the STC from...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Geneva Aviation Master Drawing List, doc. ref. GA156, Revision C, dated April 14, 2008. FAA approved 17, 2008.</td>
<td>This STC provides approval for structural provisions only. A separate approval is required for the installation of any equipment in the supplemental avionics console. The installation of this modification by third persons is subject to written permission of the approval holder and holding and disposal of...</td>
<td>06/06/2016</td>
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<td>DART AEROSPACE Ltd.</td>
<td>FAA STC</td>
<td>AS355F2, AS355 N, AS355 NP, AS355 E, AS355 F, AS355 F1</td>
<td>Installation of an air ambulance unit</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Drawing List GA178 Rev. B dated March 21, 2008.</td>
<td>Increase in the maximum climb performance of the model AS355F2.</td>
<td>01/02/2012</td>
<td>Active</td>
<td>10016984</td>
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<td>SPECTRUM AEROMED, LLC b/t SPECTRUM AEROMED USA, INC.</td>
<td>FAA STC</td>
<td>SR00388SE</td>
<td>Installation of an air ambulance unit</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Drawing List GA178 Rev. C dated February 20, 2008 and Amendment 27-23 dated February 14, 2008.</td>
<td>Increase in the maximum climb performance of the model AS355F2.</td>
<td>07/10/2014</td>
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<td>EAGLE COPTERS USA, INC.</td>
<td>FAA STC</td>
<td>SH94-14</td>
<td>Installation of an air ambulance unit</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Drawing List GA178 Rev. B dated March 21, 2008.</td>
<td>Increase in the maximum climb performance of the model AS355F2.</td>
<td>14/10/2015</td>
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<td>EAGLE COPTERS USA, INC.</td>
<td>FAA STC</td>
<td>SR00361SE</td>
<td>Utility Rack Installation per MIL-G-1644 Notes: 1. The STC was initially issued to Geneva Aviation, Inc. (initial reference: EASA.IM.R.S.01413). 2. Revision 1 to this STC reflects modification to the installation of the utility rack. 3. The STC reflects the transfer of the original product to EASA.IM.R.S.01413 from Geneva Aviation, Inc. to Eagle Coplers USA, Inc.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Drawing List GA164 Rev. B dated May 16, 2008.</td>
<td>Increase in the maximum climb performance of the model AS355F2.</td>
<td>03/07/2016</td>
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<td>10017002</td>
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<td>DARTEC AEROSPACE USA, Inc.</td>
<td>10017003</td>
<td>REV. 1</td>
<td>FAA STC SH4785NM</td>
<td>AS350B, AS350B1, AS350B2, AS350B3, AS350B4, AS350D</td>
<td>Installation of replacement dome lightvent assembly. This STC is a validation of FAA STC SH4785NM.</td>
<td>The Certification Basis for the original product remains applicable to FAA STC SH4785NM. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>General Aviation Master Drawing List, doc. ref. GA123, Revision B, dated September 22, 2011</td>
<td>11/09/2004 or later approved revision - Instructions for Continued Airworthiness, doc. ref. IDL 00771 revision dated 11/09/2004 or later approved revision - Installation Drawing L...</td>
<td>15/02/2013</td>
<td>Active</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>10017018</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>EUROCOPTER FRANCE AS 36943 EUROCOPTER FRANCE AS 36942 EUROCOPTER FRANCE AS 36941</td>
<td>Installation of a Honeywell Chadwick Model 15175 VXP System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. FM900771 revision - dated 1/11/2005 or later approved revision - Flight Manual Supplement 2, ICAO771 revision dated 11/09/2004 or later approved revision - Instructions for Continued Airworthiness 3, IDL 00771 revision dated 11/09/2004 or later approved revision - Installation Drawing L...</td>
<td>none</td>
<td>29/06/2011</td>
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<tr>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>10017020</td>
<td>REV. 3</td>
<td>FAA STC SRI01685SE</td>
<td>R44, R44 II</td>
<td>Cargo Hook System Kit. The installation as defined in MDL 155-124-00 provides for differential optional Cargo Hook System kits depending on aircraft voltage system, with and without load weighing system and the respective cargo hook inaccordance with FAA STC SRI01685SE. With revision 2 of EASA STC 10017020...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Drawing List (MDL) No. 155-124-00, Revision 19, dated 31 March 2017, or later FAA approved revision 2. Continued Airworthiness (ICA), Document No. 123-030-00, Revision 19, dated 28 March 2017, or later FAA approved revision 3. Robinson Flight Manual Supplements, Document No. 121-048-00, Revision 3.</td>
<td>Approval of this change in type design applies only to those Robinson helicopter models listed above which are equipped with Robinson hardpoints, with or without stock accessories. Port Number (PN) D134-1 Cargo Hook Kits models 200-324-00 and 200-326-00 are eligible for installation on those helicopters with 14 volt system...</td>
<td>08/05/2018</td>
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<tr>
<td>BLR AEROSPACE, L.L.C.</td>
<td>10017028</td>
<td>REV. 1</td>
<td>FAA STC SRI01685SE</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of dual tailboom strakes. Revision 1 updates the STC holder’s address</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>See EASA Approved Model List</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/05/2019</td>
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<tr>
<td>NORTHEAST ENGINEERING &amp; DEVELOPMENT TD.</td>
<td>10017030</td>
<td>REV. 1</td>
<td>TCCA STC SH97-59</td>
<td>R44, R44i, R66</td>
<td>Bear Paw Installation. With revision 1 of this STC the R66 is added as applicable type/model.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>For initial revision (Bear Paw installation for R44, R44i). Master Drawing List 96051 Issue 2, dated 04 November 2004 - Instructions for Continued Airworthiness 96051CA Issue 2, dated 16 January 2008 For revision 1 (Bear Paw installation for R66). Master Drawing List 96051 Issue 4, dated 27 July 20...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/03/2018</td>
<td>Active</td>
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<tr>
<td>AIRFILM CAMERA SYSTEMS, INC.</td>
<td>FAA STC SR02138LA</td>
<td>(EASA.IM.R.S.01489)</td>
<td>REV. 1</td>
<td>Installation of Utility Step Mount</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>1. Airfilm Camera Systems Master Drawing List No. AF135-010, Revision A, dated 3 May 2008.2. Installation Manual No. AF135-006 dated 1 May 2008.3. Instructions for Continued Airworthiness AF135-008, dated 10 March 2008.4. Flight Manual Supplement No. Document Number RFMS135-007, dated 24 April 2008.</td>
<td>Prior to installation of this design change it must be determined whether the configuration of the fuselage has been modified (afurther increase in MTOW).</td>
<td>17/09/2011</td>
<td>Active</td>
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<tr>
<td>DART AEROSPAC... USA, Inc.</td>
<td>FAA STC SR01853SE</td>
<td>(EASA.IM.R.S.01489)</td>
<td>REV. 1</td>
<td>Skidtube Installation ref. D117-762 Rev. 1: Eligibility extended to MBB-BK117 D-2</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- Master Drawing list (MDL) MBL-04177-862, Revision L, dated 23 May 2019, or later Federal Aviation Administration (FAA) approved revision. - Installation Instructions for Continued Airworthiness dated 13 July 2019, or later FAA approved revision. - Instructions for Continued Airworthiness (CA) CA-117-7...</td>
<td>This approval is allowed with Airbus MBB-BK 117 D-2, Mod. 4444 with increase MTOW to 3800Kgs. Prior to installation of this design change it must be determined whether the configuration of the fuselage has been modified (afurther increase in MTOW).</td>
<td>18/11/2019</td>
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<td>MAX-VIZ Inc.</td>
<td>FAA STC SR06532DE</td>
<td>(EASA.IM.R.S.01500)</td>
<td>REV. 1</td>
<td>Enhanced Vision Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>1. DL2/214212EV-100 revision 0 dated 11 July 2008 or later EASA approved revision. 2. Master Data List 2. PR-214212EV-120M revision 3 dated 7 May 2007 or later EASA approved revision. 3. Instructions for Continued Airworthiness dated 13 July 2007 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined whether the configuration of the fuselage has been modified (afurther increase in MTOW).</td>
<td>01/02/2018</td>
<td>Active</td>
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<td>MAX-VIZ Inc.</td>
<td>FAA STC SR06532DE</td>
<td>(EASA.IM.R.S.01501)</td>
<td>REV. 1</td>
<td>Enhanced Vision Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>1. DL-206/407EV-100 revision 0 dated 10 April 2006 or later EASA approved revision. 2. Master Data List 2. PR-206/407EV-100M revision 0 dated 25 July 2007 or later EASA approved revision. 3. Instructions for Continued Airworthiness dated 10 October 2007 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined whether the configuration of the fuselage has been modified (afurther increase in MTOW).</td>
<td>01/02/2018</td>
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<tr>
<td>10017057</td>
<td>FARGO MANUFACTURING CO.</td>
<td>REV. 2</td>
<td>FAA SR01813CH</td>
<td>MD600</td>
<td>Installation of Auxiliary Fuel System Kit.</td>
<td>- Fargo Installation Instructions Doc No. A-900-324, Rev A, dated 24th April 2008 - EASA approved Flight Manual Supplement. EASA Doc No.990, dated 24 November 2008 Or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical ImplementationProcedures.</td>
<td>Prior to installation of this changes/repair it must be determined that the interrelationship between this changes/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/04/2020</td>
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<td>10017071</td>
<td>L-3 WESCAM INC.</td>
<td>REV. 1</td>
<td>TCCA STC SH04-29</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>MX-Series Camera Installation (Structural Provisions Only).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged. The requirements for electrical aspects will require documentation to reflect the provisions of the current Fللحة.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
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<td>10017072</td>
<td>DONALDSON COMPANY, Inc.</td>
<td>REV. 2</td>
<td>FAA STC SR02383CH</td>
<td>AS350 B, AS350 B1, AS350 B2, AS350 B3, AS350 BA</td>
<td>Installation of Inlet Barrier Filter system.Revision 1 reflects the name and address change from AerospaceFiltration Systems, Inc. to Donaldson Company, Inc. Revision 2 is a change to the process for acceptance of revision to a technical document to reflect the provisions of the current EASA IM. R/S.01501(OMNIAntenna) and a MX-10 Camera applicability.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged.</td>
<td>24/08/2021</td>
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<td>DART AEROSPACE Ltd.</td>
<td>REV. 1</td>
<td>TCCA STC SH91-20</td>
<td>A335F2, AS350F, AS350NF, AS350F1, AS350D, AS350D1, AS350B3, AS350B3A, AS350C, AS350B1, AS350B2</td>
<td>D350-561 Cargo Mirror installations.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for electrical aspects will require documentation to reflect the provisions of the current Fللحة.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged.</td>
<td>03/09/2010</td>
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<td>10017077</td>
<td>DART AEROSPACE Ltd.</td>
<td>REV. 2</td>
<td>TCCA STC SH92-16</td>
<td>3. 214B, 214B-1, 2. 212, 412, 412EP 1. 204B, 205A-1</td>
<td>D212-580 Cargo Mirror/LED Pulse Light Installation. This revision adds the LED Pulse Light installation to the Cargo Mirrors.</td>
<td>FAR 20 up to and including Amendment 29-51 for the affecteedata. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<td>26/11/2010</td>
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AEROLITE AG
REV. 3
EC135 T2 (FSP/D), EC135 T1 (FSP/D), EC135 T2+ (EC135 T2), EC135 P3 (FSP/D), EC135 P1 (FSP), EC135 P1 (CD), EC135 P2 (CD)
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10017088
AEROLITE AG
ASSOCIATED TECHNICAL TCCA
10017087
Rev.
Models
REV. 8
Status
Active
10017089
AEROSPACE Ltd.
ASSOCIATED TECHNICAL TCCA
10017087
Rev.
Models
REV. 3
Status
Active
10017086
ASSOCIATED TECHNICAL TCCA
10017088
Rev.
Models
REV. 3
_status
Active

List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<th>System Record</th>
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<tr>
<td>BABCOCK M ISSION CRITICAL SERVICES F M ANAGEMENT S.A.U.</td>
<td>10017093</td>
<td>REV. 2</td>
<td>-</td>
<td>-</td>
<td>Installation of FLIR Ultra Force II System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Doc. CCM021 &quot;Documento Técnico&quot;, Ed. 1, dated 3 July 2004; Doc. No. CMA021-ID “Informe sobre el Diseño”, Ed. 1; Installation Instructions in accordance with - Doc. No. CMA021-ID “Boletín de Ingeniería”, Edición 1; Doc. No. CMA021-MID “Instrucciones de Mantenimiento e Instalación” Ed. 1. UltraForce.</td>
<td>Subjected to the conditions, limitations and procedures contained in Docho. CMA021- AFMS “Supplement for FLIR System - Ultra Force II”, Iss. 1Rev. 0 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/02/2019</td>
<td>Active</td>
<td>10017102</td>
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<tr>
<td>AVIATION SUPPORT GmbH</td>
<td>10017094</td>
<td>REV. 1</td>
<td>BO105 S (CBS-4, CBS-5)</td>
<td>-</td>
<td>Installation of CHARM I - Gas Detection System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS27, Amendment 41/1/5. (not required by Part 21,101.1 27 - 21, 25, 27, 32, 31, 303, 305, 307, 309, 311, 562, 60.</td>
<td>- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/06/2017</td>
<td>Active</td>
<td>10017104</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
<td>10017100</td>
<td>REV. 1</td>
<td>HUGHES 360D, HUGHES 360E</td>
<td>-</td>
<td>Installation of Isolator Washing Platform Ref. 1 Transfer to ACC Columbia Jet Service GmbH</td>
<td>- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and any other pre-existing design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>- Flight Manual Supplement SPOH, 12155, dated 31.07.2006 or later EASA approved revision-Umrüstungsanweisung (Installation Manual) II, 12155</td>
<td>24/02/2022</td>
<td>Active</td>
<td>10017110</td>
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<tr>
<td>HELI AUSTRIA GmbH</td>
<td>10017101</td>
<td>REV. 3</td>
<td>MD900</td>
<td>-</td>
<td>Upgrading of existing Single Cargo Hook to Dual Hook System Rev. 1: Transfer of STC to GVM Aerospace (Ireland) Ltd., Pembroke Place 10, D02 EDN District Dublin - Transfer of STC from Nova System &amp; Engineering Ltd previously GVM Aerospace (Ireland) Ltd. to Heli Austria GmbH, Hulptort, 560209. Johannesburg</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS27, Initial Issue: 27.581, 27.685, 27.771, 27.773, 27.777, 27.1301, 27.1309, 27.1321, 27.1351, 27.1357, 27.1391, 27.1393, 27.1389, 27.1386, 27.1387, 27.1388.</td>
<td>- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>- Flight Manual Supplement SPOH, 12155, dated 31.07.2006 or later EASA approved revision-Umrüstungsanweisung (Installation Manual) II, 12155</td>
<td>02/02/2023</td>
<td>Active</td>
<td>10017111</td>
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<td>AIRLIFT A/S</td>
<td>10017102</td>
<td>REV. 1</td>
<td>SA 365 C, SA 365 C1, SA 365 C2, SA 365 C3, SA 365 N, SA 365 N1, AS 365 N2, AS 365 N3</td>
<td>-</td>
<td>EMS-Installation/Installation of Medical Equipment and associated attachment and supports. Note: this STC is issued for the purpose of transferring STC No.EASA.R.S.01072 of Lufttransport AS to the above identified maintainer.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Flight Manual Supplement - EMS Installation 12155-05, 212-001 N LT36525-002, In-Combi Transport System LF02 Lateral Configuration 555-25,200-100N LT36525-029, 5V A Stretcher 555-25,201-100 NYLT 36525-200 or later revisions of the above listed documents approved by EASA/Technical Order N 365-25-0.</td>
<td>none</td>
<td>23/08/2012</td>
<td>Active</td>
<td>10017112</td>
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<tr>
<td>BABCOCK MISSION CRITICAL SERVICES F M ANAGEMENT S.A.U.</td>
<td>10017113</td>
<td>REV. 1</td>
<td>2, 212, 412, 412 EP 1, A8412</td>
<td>-</td>
<td>Installation of a pitot heater failure warning indicator Rev. 1: Transfer of STC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Associated Technical Documentation listed on Technical Document CMA077Issue 1 Rev. 0 Installation / Instructions for Continued Airworthiness in accordance with - Engineering Bulletin CMA077-B1 Issue 1 Rev. 0(0) Maintenance Instructions CMA077-M Issue 1 Rev. 0(0) Drawing CMA077-01-01-00 rev. 0...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/02/2019</td>
<td>Active</td>
<td>10017113</td>
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<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U.</td>
<td>REV. 1</td>
<td>2  212, 412, 412EP 1  AB412</td>
<td>Installation of a standby horizon emergency power supply warning/indicator Rev. 1. Transfer of STC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Associated STC Documentation listed on Technical Document CMA078 Issue 1 Rev. 0 Installation / Instructions for Continued Amortization in accordance with Engineering Bulletin CMA078-B1 Issue 1 Rev. 0 or (if applicable) Drawing CMA076-01-08-00 rev. 0, and/or operation in accordance with Flight Manual Supplement...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/02/2019</td>
<td>Active</td>
<td>10017114</td>
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<tr>
<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U.</td>
<td>REV. 1</td>
<td>S-61N</td>
<td>External host P/N 76378-260-D replacement by P/N 67378-500 Rev. 1. Transfer of STC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>CMA005-APM Supplement for Air Equipment Electric Heist Goodrich Pl. N67378-500 CMA095-MD Maintenance and installation/Removal Manual or later revisions of the above listed document(s) approved acceptability the EASA system.</td>
<td>Silosky S-61N operated by HELICJA (SN 61765, 61758, 61390, 61381 &amp; 61741). Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/02/2019</td>
<td>Surrrendered</td>
<td>10017118</td>
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<tr>
<td>MECÄER AVIATION GROUP S.p.a.</td>
<td>REV. 1</td>
<td>AB139/ AW139</td>
<td>AB139 VIF 28 P/N 6AB1KT003-001 Interior Configuration. This STC supersedes EASA STC EASA.R.S. 01219.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement No 6AB1RFM-1 Approval of this change in type design applies only to those AB139-rotorcraft previously equipped with EASA approved Installation of AgustaKIT VIP PROVISION (P/N 420500F0011).</td>
<td>09/06/2020</td>
<td>Active</td>
<td>10017129</td>
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<tr>
<td>MECÄER AVIATION GROUP S.p.a.</td>
<td>REV. 5</td>
<td>AB139, AW139</td>
<td>VIP 29 Introlors P/N 6AB1KT004-001 The revision 5 introduces eleven passenger configuration(s) (i.e. aeration of the cabin extinguisher positions and installation of newceareta.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Design Data Sheet Report DD2-2019-007 Issued Supplement Flight Manual 6AB1RFM-1.2 revision ALOwner's Manual (ICA) 6AB1WMA-2.2, Reissue 2 Rev. ACC or later revisions of the above listed document(s) approved acceptability the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/11/2019</td>
<td>Active</td>
<td>10017130</td>
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<tr>
<td>BRISTOW HELICOPTERS Ltd</td>
<td>REV. 1</td>
<td>AS 332 L, AS 332 L1</td>
<td>Rockwell Collins TCAS II, TCAS 4000 System software upgrade from Version 7.0 to 7.1. Rev 1 covers modification of Rockwell Collins TCAS 4000 TCAS II upgrade version 7.0 software. Modification MOD:332/34.5538</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- Briostow AS 332LL1 Flight Manual Supplement No. 35-Rockwell CollinsTCAS II, issue 4-1, Flight Manual Supplement No. 35-Rockwell CollinsTCAS II Version 7.1. Initial Issue or later revisions of the above listed documents approved by EASA. Certification Plan for MOD 332/34/34538...</td>
<td>Prior to installation of this design change it must be determined that the change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/08/2013</td>
<td>Active</td>
<td>10017131</td>
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<tr>
<td>RTERESSEAU DE TRANSPORT D'ÉLECTRICITÉ</td>
<td>REV. 1</td>
<td>AS 332 C1</td>
<td>Installation for Human External Cargo (HEC) applicationsNote: STC Rev. 1 is re-issuance of the original STC certificate EASA R.S. 01229 and approves compatibility of the STC with AS 332 C helicopters when equipped with the following optional equipment: AirbusHelicopters avionics suite AHCAS and Multi-p...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: JAR 29 Amdt. 3, requirement 29.21, 29.25, 29.27, 29.29, 29.49, 29.51 (c), 29.55, 29.84,29.65, 29.67, 29.75, 29.141, 29.143, 2...</td>
<td>- RTE. Master Drawing List, report no. NT-MAINLN-CNER-07H-PAIM-05-01270, inst. 3.0- RTE. Instructions Techniques de Montage*, report no. NT-MAINLN-CNER-07H-PAIM-05-01263, inst. 3.0- RTE. Manuel de Maintenance*, report no. DT.ETD-CNERS-07H-PAIM-05-01430, inst. 6.0- RTE. Supplement au VIF, report...</td>
<td>Installation of this STC requires that the helicopter is equipped with Airbus Helicopters optional equipment for Transport of Exportable Load(450kg) – Sup. 10.8 or BUP.20 as applicable to helicopter configuration. Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/09/2019</td>
<td>Active</td>
<td>10017132</td>
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<tr>
<td>STC Holder</td>
<td>STC No.</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
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<tr>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>10017136</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>EC135 T1, EC135 T2, EC135 T2+ EC135 P1, EC135 P2+</td>
<td>Change to existing EASA STC EASA.R.S.01230 - NVG Compatible Lighting System - to extend the approved Night Vision Goggles to allow night pilot operations and to include the latest EC135 types.</td>
<td>As per EC135 TCDSThe Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Modification of EASA STC EASA.R.S.01252 - NVG Compatible Lighting System.</td>
<td>None</td>
<td>20/11/2009</td>
<td>Active</td>
<td>1017132</td>
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<tr>
<td>MECANER AVIATION GROUP S.p.a.</td>
<td>10017140</td>
<td>REV. 1</td>
<td>AB139/ AW139</td>
<td>AS 350 B3</td>
<td>Bubble Window RH Installation P/N 030603/350BW according FOCA Data Sheet 56-30-01. Installation of P/N 030603/350BW according to FOCA data sheet 56-30-01.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>- 6A10AM-2-1 Rev. &quot;Owners manual&quot;</td>
<td>None</td>
<td>09/04/2010</td>
<td>Active</td>
<td>1017140</td>
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<tr>
<td>SWISS HELICOPTER ENGINEERING AG</td>
<td>10017148</td>
<td>REV. 2</td>
<td>FOCA STC 56-30-01</td>
<td>Bubble Window RH Installation P/N 030603/350BW according FOCA Data Sheet 56-30-01. Installation of P/N 030603/350BW according to FOCA data sheet 56-30-01.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>- 6A10AM-2-1 Rev. &quot;Owners manual&quot;</td>
<td>None</td>
<td>22/11/2023</td>
<td>Active</td>
<td>1017148</td>
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<tr>
<td>MECANER AVIATION GROUP S.p.a.</td>
<td>10017149</td>
<td>REV. 2</td>
<td>A1095, AW1095P</td>
<td>Cabinet Installation P/N 6A10A001-001.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Report 6A10AD-002 Rev. B &quot;Owners manual&quot;</td>
<td>None</td>
<td>01/03/2011</td>
<td>Active</td>
<td>1017149</td>
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<td>MECANER AVIATION GROUP S.p.a.</td>
<td>10017150</td>
<td>REV. 1</td>
<td>AB139/ AW139</td>
<td>CD Player Installation P/N 6AB1EL107-001.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>None</td>
<td>None</td>
<td>09/06/2010</td>
<td>Active</td>
<td>1017150</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<td>10017153</td>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>260D</td>
<td>REV. 1</td>
<td>260D</td>
<td>01261</td>
<td>Introduction of EAC-033-000 Airborne Camera SystemRev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product/changes applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Aviation Traders Modification ATL/335K003 &amp; Definition Document n°ID/O019700 dated 14 March 2006Aviation Traders RFM Supplement Number 042 Issue 1 dated September 15th later revisions of the above listed document(s) approved.</td>
<td>None</td>
<td>09/11/2012</td>
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<td>10017155</td>
<td>PREMIAR AIRCRAFT ENGINEERING LIMITEDEUROPE S.A.S.</td>
<td>269D</td>
<td>REV. 2</td>
<td>269D</td>
<td>01261</td>
<td>Installation of Honeywell KXX EGPS/NS Note: This revision is to amend an existing EASA STC (RNP-EASA.R.S. 01261 rev. 1) to add the AS 365N3 model</td>
<td>The Certification Basis for the original product stated in TCDS N. 15 parameters applicable to this certificate/ approval</td>
<td>-Installation in accordance with PremiAR Aircraft Engineering Modification reference 09-026 issue 2.- PremiAR Aircraft Engineering RFM Supplement 09- AS365N3-03 issue 1.- Installation in accordance with PremiAR Aircraft Engineering Modification reference 06-013-PremiAR Aircraft Engineering RFM Suppl.</td>
<td>None</td>
<td>01/02/2010</td>
<td>Active</td>
<td>10017155</td>
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<tr>
<td>10017156</td>
<td>ZODIAC AEROSAFETY SYSTEMS</td>
<td>260D</td>
<td>REV. 12</td>
<td>260D</td>
<td>01261</td>
<td>Installation of AERAZUR Internal Auxiliary Fuel System (IAFS) connector(s) p/n 506817-1 or 506817-2. 3 STC revision 1 introduces kit p/n 506817-1 and revised documentation (STC revision 2) introduces kit p/n 506817-2 and revis...</td>
<td>The Certification Basis for the original product and the following add-on or alternative alternative/without requirements are applicable to this certificate/ approval. Special Conditions to include factors/ for CRI E-1 issue 3- Fuel system crash resistance - Rearw...</td>
<td>S-92A Flight Manual Supplement n° 0605437, edition 16 dated 06 December 2010, and- AERAZUR IAFS Completion Kit p/n 50492-02 or 50492-03 approved under EASA STC n° 10017259 revision 3 or later. When including...</td>
<td>The IAFS Completion Kit can be installed only on S-92A helicopter equipped with:- S-92A Avionics Management System (AMS) version n° 091-2527-07 or later approved version, and- AERAZUR IAFS Provision Kit p/n 50492-02 or 50492-03 approved under EASA STC n° 10017259 revision 3 or later. When including...</td>
<td>None</td>
<td>08/01/2015</td>
<td>Active</td>
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<tr>
<td>10017157</td>
<td>AEROLITE AG</td>
<td>EM5 Equipment P/N 145221-501...This revision 1 to STC EASA R.S. 01265 introduces modification on existing cabin configurations. See MDL-06-008 Rev. C.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirement for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>EASA STC no. 01270. supersedes EASA STC EASA.R.S. 01265.</td>
<td>None</td>
<td>EASA STC no. 01270. supersedes EASA STC EASA.R.S. 01265.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>FMS-05-005 Rev. B. Chapter 04 (Aerowithness Limitations) of OPM-06-007 Rev. B. prior to installation of this design change it must be determined that the interrelationship between this design change/repair and other previously installed design change/ repair will introduce no adverse effect upon the aerowithness of the product.</td>
<td>None</td>
<td>25/10/2016</td>
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<td>10017157</td>
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<tr>
<td>10017161</td>
<td>MECAR AVATION GROUP S.p.A.</td>
<td>AB138/AW139</td>
<td>REV. 1</td>
<td>260D</td>
<td>01261</td>
<td>Cabinet Utility Installation P/N 6AB150D-001. This STC supercedes EASA STC EASA.R.S. 01269.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/ approval.</td>
<td>NOCX-002. Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>- NDC n° 6AB150D-001.- Utility Cabinet Installation Owner's Manual, document n° 6AB1WMM-2-6</td>
<td>Cabinet Utility Installation P/N 6AB150D-001. is compatible with Agustaw. P.A. cabin arrangements n° 3GS250F00111 and 3GS250F00211.</td>
<td>None</td>
<td>09/06/2010</td>
</tr>
<tr>
<td>10017162</td>
<td>MECAR AVATION GROUP S.p.A.</td>
<td>AB138/AW139</td>
<td>REV. 1</td>
<td>260D</td>
<td>01261</td>
<td>115V AC Power Installation Electrical power connectors installation in the passenger compartment. This STC supercedes EASA STC EASA.R.S. 01269.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>- Technical Fix per S.E.I DC n° 6AB1EL-004/ III - Drawing n° 6AB1EL035-001, Owner’s Manual n° 6AB1WMM-2-9</td>
<td>None</td>
<td>09/06/2010</td>
<td>Active</td>
<td>10017162</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>MECAR AVIATION GROUP S.p.A.</td>
<td>REV. 1</td>
<td>AB133/ AW139</td>
<td>ICS Installation</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The certification therefore or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- Technical Data S.E.NDC n. 6AB1EL-003/03 (Drawing n. 6AB1EL-003-001-Owner’s Manual n. 6AB1WMN-2-10)</td>
<td>None.</td>
<td>09/02/2010</td>
<td>Active</td>
<td>1001711C</td>
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<td>MECAR AVIATION GROUP S.p.A.</td>
<td>REV. 3</td>
<td>AB133/ AW139</td>
<td>VIP 43 Interiors</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- RFD Supplement VIP 43 Interiors 6AB1RFM-1-4. Rev. 1.2. Technical Data of the above listed documents approved by EASA. Owner’s Manual 6AB1WMN-2-10-3, Rev. 2.</td>
<td>Prior to installation of this change it must be determined that the installation of Honeywell KTA 870 Traffic &amp; Emergency Medical Intercom, EMS, SAT-COM headsets, small cabinet and other minor items will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/04/2015</td>
<td>Active</td>
<td>1001711C</td>
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<td>KUERZI AVIONICS AG</td>
<td>REV. 3</td>
<td>SA 330 J</td>
<td>Avionics UpgradeGPS.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document and Drawing List No. 6AB1WMN-21059 Rev. C or later. Approved revisions &amp; Flight Manual Supplement Garmin GPS150 No. FMS-200616559 Rev. A or later. Approved revisions</td>
<td>Prior to installation of this change/repair it must be determined that the installation of Honeywell KTA 870 Traffic &amp; Emergency Medical Intercom, EMS, SAT-COM headsets, small cabinet and other minor items will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/06/2016</td>
<td>Active</td>
<td>1001711C</td>
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<td>AEROLITE AG</td>
<td>REV. 2</td>
<td>EC135 T3</td>
<td>Installation of</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Rotorcraft Flight Manual Supplement FMS-210 Revision E applicable to P/N 135050-501.</td>
<td>The initial issue of this STC supersedes EASA STC 2004-242 and Major Modification No. 2004-11064. Limited to helicopters with CDS/CPS cockpit variant. Extension of applicability to EC135 T3/CPDS.</td>
<td>06/07/2022</td>
<td>Active</td>
<td>1001711C</td>
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<td>ECOMS AVIATION SYSTEMS GMBH</td>
<td>REV. 1</td>
<td>EC635 T3(CPDS)</td>
<td>Emergency Medical</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement BWS02, fms, issue 01-07-2007 or later.</td>
<td>The initial issue of this STC is limited to rotorcraft S/N 5480 Prior to installation of this STC.</td>
<td>11/05/2015</td>
<td>Active</td>
<td>1001711C</td>
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<tr>
<td>PREMIAIR AIRCRAFT ENGINEERING LIMITED</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBILITY</td>
<td>Installation of</td>
<td>The Certification Basis for the original product stated in TCDS N. 15 references applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement BWS02, fms, issue 01-07-2007 or later.</td>
<td>The initial issue of this STC is limited to rotorcraft S/N 5480 Prior to installation of this STC.</td>
<td>11/05/2015</td>
<td>Active</td>
<td>1001711C</td>
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For more information, visit [EASA.europa.eu](https://easa.europa.eu)
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<th>STC Holder</th>
<th>Associated Technical</th>
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<th>Associated Technical Documentation</th>
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<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>PREMIAR AIRCRAFT ENGINEERING LTD</td>
<td>CA-AVR UN RESPONSIBILI TY</td>
<td>Installation of Honeywell RMD 350 Multi Function Display/Nav. This revision is to amend an existing EASA STC (Ref: EASA.R.S.01290 rev. 1) to add the AS 365N3 model</td>
<td>The Certification Basis for the original product states in TCDS N. 156 remains applicable to this STC/approval.</td>
<td>-Installation in accordance with PremiAir Aircraft Engineering Modification reference 09-024 issue 2 -PremiAir Aircraft Engineering RRM Supplement 09- AS365N2-01 Issue 1 -Installation in accordance with PremiAir Aircraft Engineering Modification reference 07-008 PremiAir Aircraft Engineering RRM Supplement...</td>
<td>None</td>
<td>01/02/2010</td>
<td>Active</td>
<td>1001717</td>
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<td>MECAR AVIATION GROUP S.p.A.</td>
<td>VIP S5 Interiors PN 6AB1AD007-001</td>
<td>The change consists in customizing passenger cabin interiors/personalization seat layout, carpet and liners, devices, adding a small cabinet and other minor items. This STC supersedes EASA STC EASA.R.S.01294.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Publication nr. 6AB1RFM-1-S -Supplement RRM</td>
<td>The change in type design applies only to those AB139/ AW139, helicopters previously equipped with EASA &quot;Kit VIP Provision PN 402SOP01311&quot;.</td>
<td>09/06/2010</td>
<td>Active</td>
<td>1001717</td>
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<tr>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>EC135 P1/ EC135 P2/ EC135 T1</td>
<td>Revision 1 of this STC is due to the transfer of EASA STC EASA.R.S.01296. Installation of a HellFire-MIDIFPD025 Moving Map System, Revision 2, Transfer to ACC COLUMBIA Jet Service GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Rotorcraft Flight Manual Supplement P1-P2, T1 (CDS, CPDS), EC135-69008-800-S-MF Rev. 0, Instructions for Continued Airworthiness EC 135-69008-810-S-MM Rev.0.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/02/2022</td>
<td>Active</td>
<td>1001717</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
<td>MBB-BK117 B-2</td>
<td>Revision 1 of this STC is due to the transfer of EASA STC EASA.R.S.01297. Installation of a Hell Fire-MIDIFPD025 Moving Map System, Revision 2, Transfer to ACC COLUMBIA Jet Service GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Rotorcraft Flight Manual Supplement BK117 B-2, BK 117-69007-800-S-MF-2. Instructions for Continued Airworthiness BK117-69007-810-S-MM.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/02/2022</td>
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<td>1001717</td>
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<td>AEROLITE AG</td>
<td>MBB-BK 117 C-2</td>
<td>Installation of NATO Stretcher Retainer. This STC revision is issued for the purpose of transferring the STC to Aerolite AG.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment: CS 27 Amdt. 2. The requirements for environmental protection and the associated certified noise and/or emissions levels...</td>
<td>Flight Manual Supplement FMS-06-008, Rev. N/C, dated 27.05.2020 for later revisions of the above listed documents approved by EASA.</td>
<td>The NATO Stretcher (NSN 6530-01-380-7300) referred to in this installation is not part of this STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/03/2016</td>
<td>Active</td>
<td>1001717</td>
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<tr>
<td>AIRBUS HELICOPTERS</td>
<td>EC135 T2; EC135 T2+; EC135 P2; EC135 P2+</td>
<td>Daylight Camera Installation (Ultraframe II/III) (Structural and Electrical/Provisions). Installation of a camera Ultraframe II/III HD in accordance with ECE Master Document List L653H0001E01, Rev. B/Rev. 1. Transfer of STC EASA.R.S.01297 from Airbus Helicopters España to Airbus Helicopters</td>
<td>Flight Manual Supplement &quot;Supplement for FSI UM II/III HD Gyro Stabilised Camera&quot;, FMA 11.100 Installation Instructions: Master Document List DL L653H0001E01, Rev. B/Instructions for Continued Airworthiness &quot;FMS File Manual FSI UM II/III HD&quot;, Rev. A.</td>
<td>Flight Manual Supplement FMS-06-008, Rev. N/C, dated 27.05.2020 for later revisions of the above listed documents approved by EASA.</td>
<td>The NATO Stretcher (NSN 6530-01-380-7300) referred to in this installation is not part of this STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/06/2022</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>MECANER AVIATION GROUP S.p.A.</td>
<td>VIJ Report No.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise/emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise/emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>This STC is only applicable to the AB139 and AW139 helicopters equipped with the Agusta provision for VIP Cabin Interiors P/N 402550F01511.</td>
<td>09/06/2010</td>
<td>Active</td>
<td>101171</td>
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<td>ECMS AVATION SYSTEMS GMBH</td>
<td>EC135 T2+/T3(CPDS)</td>
<td>Flight Manual Supplement for External Abseil Device EAD01 for rotorcraft EC135 and EC145, EAD01, MDL EAS A, Rev. 10 Installation Instructions for the E...</td>
<td>Flight Manual Supplement for External Abseil Device EAD01 for rotorcraft EC135 and EC145, EAD01, MDL EAS A, Rev. 10 Installation Instructions for the E...</td>
<td>Prior to installation of this design change it must be determined the relationship between this design change and any other previously installed design change and/or repair will introduce roadworthiness of the product.</td>
<td>09/06/2017</td>
<td>Active</td>
<td>101201</td>
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<tr>
<td>AIRBUS HELICOPTERS</td>
<td>EC135 T2+</td>
<td>Flight Manual Supplement Supplement for Forward Looking Infrared Wescam System UK-157 FMA 11.101Installation instructions Master Document List DL LB54H0005E1 Rev. B and DL LB54H0015E1 Rev. B. Installation of the above listed document(s) approved/accep...</td>
<td>Flight Manual Supplement Supplement for Forward Looking Infrared Wescam System UK-157 FMA 11.101Installation instructions Master Document List DL LB54H0005E1 Rev. B and DL LB54H0015E1 Rev. B. Installation of the above listed document(s) approved/accep...</td>
<td>Prior to installation of this design change/repair it must be determined the relationship between this change/repair and any other previously installed change and/or repair will introduce roadworthiness of the product.</td>
<td>09/06/2022</td>
<td>Active</td>
<td>101202</td>
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<tr>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>MBB-BK117 B-2/ MBB-BK117 C-1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise/emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise/emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement Eurocopter BK117B-2/1ACC-BK117S-630-S-MF Rev. 1, dated 31.05.2007 Instructions for Continued Airworthiness ACC-BK117S-630-S-M Rev.0 dated 03.04.2007</td>
<td>24/02/2012</td>
<td>Active</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10017217</td>
<td>AEROLITE AG</td>
<td>REV. 3</td>
<td></td>
<td>EC 135 T2 (CPD6)</td>
<td>Installation of CAS CPT 900 ELT. The Certification Basis for the original product remains applicable to this certificate/approval. The certification notes/possible emissions levels of the original product unchanged and remain applicable.</td>
<td>Flight Manual Supplementation FMS 06-009 rev 04 later revisions of the above listed documents approved by EASA.</td>
<td>S/Ns installations P/N 135625-501 to -504 for Valid only for D-HEDC S/N 4197.</td>
<td>12/02/2019</td>
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<td>10017221</td>
<td>KUERZI AVIONICS AG</td>
<td>REV. 1</td>
<td>BELL 206B JET RANGER III</td>
<td>SeLA II and Archer Z-150 System Installation The present major change approval updates the existing HELLAS system installation (STC EASA R S. 01382) with a helmet heater-mounted display system and sight system (Archer Z-150) and installation of avionics-based landing aid (SeLA II).</td>
<td>Flight Manual Supplement FMS-200717881 Rev. C 19/07/2014 later revisions of the above listed documents approved by EASA.</td>
<td>Valid only for D-HEDC S/N 4197. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair. There will introduce no adverse effect upon the airworthiness of the product.</td>
<td>3/10/2014</td>
<td>Active</td>
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<td>10017222</td>
<td>CALEDONIAN AIRBORNE SYSTEMS LTD</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBILITY</td>
<td>AGUSTA-BELL 212 &amp; 412</td>
<td>Installation of CAS CPT 900 ELT.</td>
<td>FAR 29</td>
<td>None</td>
<td>2/10/2010</td>
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<tr>
<td>10017223</td>
<td>CALEDONIAN AIRBORNE SYSTEMS LTD</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BELL 212, BELL 412</td>
<td>Installation of CAS CPT 900 ELT</td>
<td>FAR 29</td>
<td>None</td>
<td>2/10/2010</td>
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<td>10017224</td>
<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U</td>
<td>REV. 1</td>
<td>2. 412, 412EP 1, A&amp;H12</td>
<td>CMA 135 &quot;Radio Attenuation Audio Warning Indicator&quot; Rev. 1</td>
<td>Transfer of STC</td>
<td>Technical Document CMA 135 Issue 1 Installation instructions for continued airworthiness in accordance with Engineering Bulletin CMA 135-B0 Issue 1</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/02/2019</td>
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<td>STC Holder</td>
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<td>NSE</td>
<td>10017228</td>
<td>STC Holder</td>
<td>REV. 5</td>
<td>AS 355 N, AS 355 NP AS 355 F1, AS 355 F2</td>
<td>Installation of BRITE SAVER 2 Data Monitoring SystemRevision 1 – Covered a documentary change to STCRevision 2 – Covered torque and memory management minor change.Revision 3 – Covered installation of ultrasonic air/ground switch, second possible location on the BRITE SAVER 2 on the instrument panel</td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:The following paragraph(s) at a later amendment: FAR 27/27.29 Annex 27.14, ...FAR 27/27.301 Original issueFAR 27/27.303 Original issue, ... FAR 27/27.305 Original issueFAR 27/27.</td>
<td>- Note explicative relative au (Description note) STC Minor Change Ref. 00028476AA Rev. 02 dated 30 June 2019 - BRITE SAVER2 RECORDER DEFINITION File Ref. CD-0027-02 Rev. 18 dated 04 January 2018 - COMPONENT MAINTENANCE MANUAL WITH ILLUSTRATED PARTS LIST ref. CM3031-03-01 Rev. 02 dated 14 March 2018 - Bel.</td>
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<tr>
<td>NSE</td>
<td>10017230</td>
<td>STC Holder</td>
<td>REV. 7</td>
<td>AS 350 BB, AS 350 D AS 350 B, AS 350 B1, AS 350 B2 AS 350 BA, AS 350 BB, AS 350 D AS 350 B2, AS 350 BA</td>
<td>Installation of BRITE SAVER 2 Usage Monitoring SystemRevision 1 – Included torque and memory management minor change.Revision 2 – Included addition of the possibility to install the ultrasonic air/ground switch and second source for SUBO connectors and method.Revision 3 – Included embedded software error</td>
<td></td>
<td>The Type and OSO Certification Bases (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Note explicative relative au (Description Note) STC Minor Change Ref.00028476AA Rev. 02 dated 06 June 2019 - BRITE SAVER2 RECORDER DEFINITION File Ref. CD-0027-00-EN Rev 18 dated 04 January 2018 - COMPONENT MAINTENANCE Manual with Illustrated Part List Ref. CM3031-03-01 Rev. 02 dated 14 March 2018 - BRITE SAVER 2</td>
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<td>AEROLITE AG</td>
<td>10017231</td>
<td>STC Holder</td>
<td>REV. 3</td>
<td>AB139, AW139</td>
<td>Installation PIN 139084-501 for 3 stretcher units with relevant cabin floor rails and featuring each, dedicated oxygen and electrical distribution systems and oxygen bottles</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Document List MDL-07-007 Rev. G - Instructions for Continued Airworthiness and Operations Manual OPM-07-010 rev.C - Service Bulletins SB-27-017 Rev.C - later revisions of the above listed document(s) approved and acceptable to the EASA system.</td>
</tr>
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<td>AEROLITE AG</td>
<td>10017232</td>
<td>STC Holder</td>
<td>REV. 1</td>
<td>EC635 P2+, EC635 T1, EC635 T2+ EC135 T2+ EC135 P2+ EC135 T2 EC135 P2, EC135 T1, EC135 P1</td>
<td>EMS Stretcher Installation PIN 138501-501-01 introduction of the Citrus Stretcher Model C-1 01.0-123-01 into the newly certified Citrus 1010 Stretcher Installation PIN 138507-501/502-503-504 (STC EASA.R.01406)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Flight manual supplement FMS-07-007 Revision A and later revisions of the above listed document approved by EASA.</td>
<td>To ensure patient head protection from injuries during crash, there must be at least 200mm of space between the forward end of each installed stretcher and any equipment installed forward of the stretcher.</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10017236</td>
<td>RUAG AG</td>
<td>REV. 1</td>
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<td>SA 365 N</td>
<td>EUROCOPTER FRANCE SA 365N</td>
<td>Enhanced navigation system/removed equipment - Pilot &amp; Copilot HSI 651- Trimble 2101 GPS w/ MD41 Annunciator Control Unit and MD41 CDI. KTR 908 VHF COM System w/ KS6 908 Control Unit -KEA 346 Encoding Alltime - BAUDR-fixed Encoding Alltime - KAS 297A Attitude Selection - Narco ELT 10 System - CD...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Documents List, Doc. No. 308-05001 Rev. 2, RFM, Doc. No. 308-05500 Rev. 1 for later revisions of the above listed documents approved/ accustomed to the EASA system.</td>
</tr>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 1</td>
<td>SA 365 N</td>
<td></td>
<td>AEROTEC modification IRS (installation radio de bord) n° 0928. Note for internal record: This application is to amend an existing EASA(Stc) (Ref P-EASA.R.S. 01452) to add the S/N 6147</td>
<td>FAR 20 amendment 16 The Certification Basis for the original product stated in TCDS N. 159 remains applicable to this certificate/approval The certificate/notes and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Installation of Aerodata Master Data List AE 0809 MDL Rev. 4 dated 23/03/2008. AEROTEC Master Data List AE 0928 MDL Rev. 0 dated 23/02/2010. Aerotec Flight Manual Supplement AE 0809 AFMS Rev. 1. dated 23/03/09 or later approved revision - Aerotec Flight Manual Supplement AE 0928 AFMS Rev. 0. dated 02/1.</td>
<td>This STC is limited to SA 365N S/N 6147 and can be installed only by AEROTEC technical staff in accordance with AEROTEC STC file Mod n° 0069 for S/N 6080-AEROTECSTC file Mod n° 0069 for S/N 6147.</td>
<td>01/02/2010</td>
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<tr>
<td>AEROLITE AG</td>
<td>REV. 2</td>
<td>EC135 T2, EC135 T2+</td>
<td></td>
<td>Change to Installation of EMS Equipment Aerolite P/N 135602-552/Change comprises of: - Introduction of Support Panel Assy/PN 004191-501 into DC Power System Inst/PN 135615-501. - Introduction of EMS Integrator Inst/PN 135601-503. - Introduction of Reversible Copilot Seat Inst/PN 135631-501-001.</td>
<td>JAR 27 The Certification Basis for the original product remains applicable to the certificate/approval The certificate/notes and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement FMS-55-001 Revision B. dated May 6th, 2011 or later revisions of the following documents sections approved by EASA: Master Document List MDL-08-004 Equipment List EQL-08-004 Operations Manual OPM-08-001 Revision B. dated May 6th, 2011.</td>
<td>This STC is limited to S/N 337 and above.</td>
<td>15/10/2012</td>
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<tr>
<td>MECAN AVIATION GROUP S.p.a.</td>
<td>REV. 1</td>
<td>AB139 AW139</td>
<td></td>
<td>Interiors Configuration Ip 131 P/N 6AB1AD009-001. This STC supersedes EASA STC EASA.R.S.01461.</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval The certificate/notes and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- RPM Supplement No. 6AB1RFM-1-7 Rev. A. dated 16 May 2008 or later approved revision - Owner's Manual No. 6AB1WMM-0-14 Rev. A. dated 16 May 2008.</td>
<td>Apogia provision P/N 402520703-31 must be installed.</td>
<td>08/06/2010</td>
</tr>
<tr>
<td>AERODATA AG</td>
<td>REV. 1</td>
<td>EC155 B1</td>
<td></td>
<td>Installation of Aerodata Airborne Observation Equipment in EC155 B1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/notes and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement “Integration of AOP in Maritime EC155, A/C Type EC155 B1. Aerodata Modification Ident: EC155-04/4” or later revisions of the above listed documents approved by EASA Certification Plan “CP-EC155-04-01”, Revision 1, 12/04/2015</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/02/2016</td>
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<tr>
<td>ZODIAC AEROSAFETY SYSTEMS</td>
<td>REV. 11</td>
<td>S-92A</td>
<td></td>
<td>Installation of Provision Kit p/n 509429-02 or p/n 509429-03 of Internally Fuel System (IFS) STC revisions 5 and 2 introduce minor design modifications for kit p/n 509429-02 and revised documentation STC revision 3 introduces kit p/n 509429-03 and revised documentation STC revision 4 clarify...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/notes and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Aezar Organigramme Technique ref. 509429-03 issue 3 dated 05 July 2013. Aezar Organigramme Technique ref. 509429-02 Issue 12 dated 26/04/2010. IAFS - High Profile (HP) version - Maintenance Manual Supplement for provision kit - chapter 11, 21,25,26,31 and 71. Issue 5 dated 14 Feb2014/AFS - Prov...</td>
<td>Refer to applicable installation instructions for compatibility of the IFS provision kit with S-92A cabin interior panels as stated in STC 10017259 revisions 1 to 97. Installation of this modification by third persons is subject to written permission of the approval holder. The approval holder shall f...</td>
<td>08/10/2014</td>
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<tr>
<td>MECAN AVIATION GROUP S.p.a.</td>
<td>REV. 01-47</td>
<td>AB139 AW139</td>
<td></td>
<td>AB139 AW139 Seat Armrests Installation P/N 6AB1AD011-001. This STC supersedes EASA STC EASA.R.S.01470.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificate/notes and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Owner's Manual for Armrests Installation P/N 6AB1AD011-001</td>
<td>This STC can only be installed on the 48/AH139 12-seat configuration P/N 30G32S3PO0111 that embodies seats P/N 3G2520V00131.</td>
<td>09/06/2010</td>
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### List of EASA Supplemental Type Certificates

**EASA.europa.eu**

<table>
<thead>
<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<th>Date of Issue</th>
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<tr>
<td>10017265</td>
<td>AEROLITE AG</td>
<td>REV. 4</td>
<td>AB139, AW139</td>
<td>AV 139 EMS Kit P/N 139007-501: Revision 1 relates changes - Streicher Inst. P/N 139190-501-500-506-507-511-513-51 5-517-401-603-605-607-611-813-81 5-617-Cargo Platform Inst. P/N 139191-501-Touchbook Retainer Inst. P/N 139225-501-922-GPS/GSM- Antenna...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Initial technical documentation: FMS-08-00 5 Revision NCOPM-08-005 Revision NC, dated May 19, 2008Revision 1 related technical documentation: FMS-08-00 5 Revision B, dated Oct 15, 2010</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification is not adversely affected.</td>
<td>20/12/2019</td>
<td>Active</td>
<td>10017265</td>
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**[EASA.R.S.01488]**

**VIP 37 Plus Interior Configuration P/N 8AB1A0512**

**10017267**

**MECAER AVIATION GROUP S.p.A.**

| REV. 4 | AB139/AW139 | VIP 37 Plus Interior Configuration P/N 8AB1A512 | The Certification Basis for the original product remains applicable to this certificate/approval. | Supplement to RFS doc. No. 6AB 1 RFS-1-8 Revision Ac revision 12 of the above listed documents approved by EASA- Owner’s Manual doc. No. SAB1WMM-2-16 Re-issue 1 Revision X-Change Description doc. No. 6AB1 RAD-216 Revision E- Notice of Design Change (NDC)- doc. No. 6AB1OAD-234 Issue A | Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification is not adversely affected. | 08/09/2014 | Active | 10017267 |

**2010**

**EASA.R.S.01494**

**NSE INDUSTRIES S.A.**

| REV. 2 | 2-R44, R44 II 1-R22, R22 ALPHAR, R22 BETA 1-R22 MARINER | Replacement of ECT log by NSE log/Modification of Maintenance Manual Supplement reference from “ST1-0023-BBB5” to “RMS-PN-R22-R44-EN/Modification of Data Monitoring System (DMS)/Useage Monitoring System (UMS)/Update of documentation due to new fittings and electronic components/Addition of an RF... | The Certification Basis for the original product remains applicable to this certificate/approval. | RMS-PN-PE-22-R44 Rev 01 dated 24/11/2014 or later revisions of the above listed documents approved by EASA-Conf/Reg-004-056 E6-EN Rev 05 dated 24/11/2014 | Prior to installation of this change it must be determined that the interrelationship between this change and any other previously installed modification is not adversely affected. | 04/12/2014 | Active | 10017272 |

**2010**

**EASA.R.S.01503**

**ECMS AVIATION SYSTEMS GMBH**

| REV. 1 | EC635 T1, EC635 T2*, EC635 T3, EC635 P2* | Installation of HEC Rope System HRS 01 Rev. 1 Extension of applicability to EC135 P3, EC135 T3, EC635 P3, EC635 T3, Additon and removal of components and removal of a new high load variant. Associated change to RFSMS and CMM. | The Certification Basis for the original product remains applicable to this certificate/approval. | Compliance Program, Doc. Ref. HRS01_MDC_EC135 EAS A Revision 5- Flight Manual Supplement for HEC Rope System HRS 01 on EC135, EC635, Doc. Ref. HRS01_RFMS_EC135_EA SA Revision 1- Airworthiness Limitations - Lifetime of PD52 components, Doc. Ref.PD55_Lifetime_Limit ings_EASA Revision 2- Equipment List F | Prior to installation of this change it must be determined that the interrelationship between this change and any other previously installed modification is not adversely affected. | 02/06/2019 | Active | 10017273 |

**2010**

**EASA.R.S.01524**

**AEROLITE AG**

<p>| REV. 3 | AB139, AW139 | EMS Equipment 13917-2051-502Revision 1 to incorporate PAX seat configurations Aerostar P/N139117-505, 139117-506 and 139117-507 of 12, 14 and 15 PAX seats (as per STC EASA R.S.01524) Revision 2 to introduce Ceiling Unit Assy configuration P/N 0662-458 equipped with NVG friendly cabin lig... | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. | Flight Manual Supplement FMS-08-008 Cor later revisions of the above listed documents approved by EASA- Operator’s Manual OPM-08-010 D- Master Document List MDL-08-015 D | EMS Fixed Provisions Kit P/N 139125-502 (STC EASA R.S.01474) and Aerolite SADD-12-027 must be installed as a prerequisite for configurations asdefined in applicable design data. Prior to installation of this design change it must be determined that the interrelationship between this design change and any... | 30/07/2015 | Active | 10017279 |</p>
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<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
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<th>Limitations/Conditions</th>
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<tr>
<td>AEROLITE AG</td>
<td>REV. 5</td>
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<td>MBB-BK 117 C-2</td>
<td>Styloflex Package (VIP Kit) Agusta P.N. 145305-051-502 (Revision to EASA R STC 01205)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>ECMS AVIATION SYSTEMS GMBH</td>
<td>REV. 2</td>
<td></td>
<td>MBB-BK 117 D-2M</td>
<td>Installation of HEC Rope System HR505Rev. 1: Extension of applicability to BK117 C-2, BK117 D-2 and BK117D-2XProof/Rev. 2. Addition of removal of components, introduction of new highload variant, Associated change to RFMS and CMM.</td>
<td>The Certification Basis for the original product as ameneded by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment CS29 amendment 3:29.25, 29.27, 29.65, 29.141, 29.251, 29.301, 29.303, 29.305, 29.307, 29.309, 29.337, 29.341, 29.571, 29.80.</td>
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<td>AIR AMBULANCE TECHNOLOGY GMBH</td>
<td>REV. 1</td>
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<td>AB139, AW139</td>
<td>Installation of Air Ambulance Equipment, Mission Equipment “Quick -Conversion”, P/N AW139-25-20-37.000-60x model</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CS 29 Amd1 2CS 29.1, CS 29.21, CS 29.25, CS 29.39, CS 29.39, CS 29.39, CS 29.39, CS 29.39, CS 29.39, CS 29.471, CS 29.562, CS 29.571, CS 29.601, CS 29.603, CS 29.805, C</td>
<td>Master Data List AB/AV 139 Air Ambulance Equipment Doc. No.”MDL-AB/AV 139-37, 000-60X”, Rev. 0</td>
<td>Rotorcraft Flight Manual FMS-08-009 Revision B, later revisions of the above listed documents approved by EASA. Operation Manual OPM-08-11 Revision C, Master Document List MDL-08-11 Revision E</td>
<td>When the Cargo Separation Wait is installed the original MBB-BK117 C-Corona detector kit must be installed in the cargo compartment. When operating the rotorcraft with the doors open or removed, the limitations given in Flight Manual Supplement No. FMS 9.1.2 must be followed. For Single Pilot operation???</td>
<td>15/03/2016</td>
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<td>REV. 5</td>
<td></td>
<td>MBB-BK 117 C-2</td>
<td>Styloflex Package (VIP Kit) Agusta P.N. 145305-051-502 (Revision to EASA R STC 01205)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>ECMS AVIATION SYSTEMS GMBH</td>
<td>REV. 2</td>
<td></td>
<td>MBB-BK 117 D-2M</td>
<td>Installation of HEC Rope System HR505Rev. 1: Extension of applicability to BK117 C-2, BK117 D-2 and BK117D-2XProof/Rev. 2. Addition of removal of components, introduction of new highload variant, Associated change to RFMS and CMM.</td>
<td>The Certification Basis for the original product as ameneded by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment CS29 amendment 3:29.25, 29.27, 29.65, 29.141, 29.251, 29.301, 29.303, 29.305, 29.307, 29.309, 29.337, 29.341, 29.571, 29.80.</td>
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<tr>
<td>AIR AMBULANCE TECHNOLOGY GMBH</td>
<td>REV. 1</td>
<td></td>
<td>AB139, AW139</td>
<td>Installation of Air Ambulance Equipment, Mission Equipment “Quick -Conversion”, P/N AW139-25-20-37.000-60x model</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CS 29 Amd1 2CS 29.1, CS 29.21, CS 29.25, CS 29.39, CS 29.39, CS 29.39, CS 29.39, CS 29.39, CS 29.39, CS 29.39, CS 29.39, CS 29.471, CS 29.562, CS 29.571, CS 29.601, CS 29.603, CS 29.805, C</td>
<td>Master Data List AB/AV 139 Air Ambulance Equipment Doc. No.”MDL-AB/AV 139-37, 000-60X”, Rev. 0</td>
<td>Rotorcraft Flight Manual FMS-08-009 Revision B, later revisions of the above listed documents approved by EASA. Operation Manual OPM-08-11 Revision C, Master Document List MDL-08-11 Revision E</td>
<td>When the Cargo Separation Wait is installed the original MBB-BK117 C-Corona detector kit must be installed in the cargo compartment. When operating the rotorcraft with the doors open or removed, the limitations given in Flight Manual Supplement No. FMS 9.1.2 must be followed. For Single Pilot operation???</td>
<td>15/03/2016</td>
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**Note:** The table above provides a structured representation of the EASA Supplemental Type Certificates, including STC Holder, Rev. level, Foreign/NAA STC ref., Model, STC description, Certification basis, Associated Technical Documentation, Limitations/Conditions, Date of Issue, Status, and System Record. The data is extracted and formatted to ensure clarity and readability.
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<th>STC Holder</th>
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<th>Status</th>
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<tr>
<td>AEROLITE AG</td>
<td>FOCA STC 25-20-65 EC135 T3(CPDS), EC135 T2(CPDS), EC135 T1(CPDS), T1 (CPDS), EC135 P3(CPDS), EC135 P2(CPDS), P2+, EC135 P1(CPDS), P1 (CPDS).</td>
<td>Initial issue: Climsheld Door Lock installation P/N 130656-501-502 Note: This STC is based on the FOCA STC Z 25-20-65. Its applicability extended to P2+ and T2+ models.Revision 1 to STC, revision of the applicability of the Climsheld Door Lock Installation P/N 130656-501-502 to the T3 and P3 varia...</td>
<td>The Certification Basis (CB) for the original product remains applicable; this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Initial installation Manual Doc. No. IM-901 Maintenance Manual Doc. No. 52-31-01 Revision 1 to STC CPR-15-036 Revision A Master Document List M-607 Revision DEquipment List EQL-15-013 Revision NC Master Drawing List DLW-807 Revision FixInstallation Manual IM-901 Revision FixMaintenance Manual OM-52...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2019</td>
<td>Active</td>
<td>10017292</td>
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<tr>
<td>AEROMECH Inc.</td>
<td>FAA STC SA01602SE CESSNA 441 SERIES</td>
<td>Group Approval for Reduced Vertical Separation Minimum (RVSM)</td>
<td>FAR Part 23</td>
<td>Airplane Flight Manual Supplement as per AEROMECH document: ...FM01 - for Class 1 airframes dated 03 June 2009, ...AEROMECH approved revision: ...FM02 - for Class 2 airframes, dated 06 June 2009, or later. ...AEROMECH approved revision: Instructions for continued airworthiness as per AEROMECH docu...</td>
<td>This approval is issued based on the FAA STC SA01602SE, issued on 26 October 2005 and later amended on 17 November 2006.</td>
<td>06/08/2009</td>
<td>Active</td>
<td>10026757</td>
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<tr>
<td>DSA A.S.</td>
<td>172RG</td>
<td>Avionics Upgrade - GNS430W, KX165A, GMA340, GI106A, KN62A, GTX330/KNANAD406AF Compact Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable; this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>Flight manual supplement S21 LDA04/009/C172DG-DSI 435 W VHF COMM Transceiver, VOY ILS and GPS Receiver S22 LDA 03/09/C172DGXX 165A VHF COMM TRANSCIVER NAV RECEIVER S23 LDA 03/09/C172DGXX 165A NAV indicator S24 LDA 03/09/C172DGMA 340 AUDIO PANEL S25 LDA 03/09/C172DGTX 330 S-MODE TRANSPONDERS S26 LDA 03/09/...</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse relationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/05/2020</td>
<td>Active</td>
<td>10026761</td>
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<td>B/E AEROSPACE LIMITED</td>
<td>TCCA LSTC C-LSA09-026, IS7-1 CL-600-2B19</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>The Certification Basis (CB) for the original product remains applicable; this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA LSTC No. C- LSA09-026 Issue 1 dated 04 August 2009 Aero Consulting Services Ltd. Modification Data Summary G103000 Revision G103000 Revision W.10.2007s1 Revision W.10.2007s2 Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation.</td>
<td>Restricted to AS9837L.</td>
<td>16/06/2013</td>
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<td>10026784</td>
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<td>PA-46-500TP, PA-46R-350T, PA-46-310P</td>
<td>Installation of Single or Dual Garmin G500 / G500 Avionics System(INCLUDING SVS) and Dual Garmin GNS4330/XN GPS/COMM/NAV.</td>
<td>-EASA approved Master Document Ltd, Doc. No. ASR-2009-075- MDL-01-00,Amtd. Level 00, dated 20-Jun-2010, or later approved amdt. If EASA approved Instructions for Installation, Doc. No. ASR-2009-075- IFI-10-00, Amtd. Level 00, dated 18-Jun-2009, or later approved amdt. Level -EASA approved Flight Manual...</td>
<td>This STC is limited to the configuration of aircraft, equipment type and functions as listed in EASA approved Instructions for Installation Doc. No. ASR-2009-075-IFI-10-00, Amtd. Level 00, dated 18-Jun-2009, especially as mentioned in item 2. of this document; or later approved amendment...</td>
<td>23/04/2010</td>
<td>Active</td>
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<td>10026785</td>
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<td></td>
<td>95-C55, D55, E55, 58, G58</td>
<td>Installation of 4-blade MTV-14-D-C-F-FC 188-30 propellers HawkerBeechcraft Model 95-C55, D55, E55, 58, G58.</td>
<td>FAR 23 including amendment 34</td>
<td>STC approved for Piper PA-24 series Carlisle, PA1788.</td>
<td>10/08/2009</td>
<td>Active</td>
</tr>
<tr>
<td>10026805</td>
<td></td>
<td></td>
<td>BEECH 1900D</td>
<td>Installation of Skytrac ISAT-100 GPS/Iridium communicator system</td>
<td>As defined in the FAA TCDS A24CE -Design Order Package DFE-03-0381 DOP-00, dated 15-05-09-Declaration of Compliance, dated 11-07-09</td>
<td>None.</td>
<td>11/06/2009</td>
<td>Active</td>
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<tr>
<td>10026811</td>
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<td>BOEING 767-300</td>
<td>Installation of LIFEPdot Stretchers.</td>
<td>As per TCDS.</td>
<td>STC approved for Piper PA-24 series Carlisle, PA1788.</td>
<td>12/08/2009</td>
<td>Active</td>
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<tr>
<td>10026831</td>
<td></td>
<td></td>
<td>PA-24, PA-24-250, PA-24-260, PA-24-400</td>
<td>Approval of an Aspen EFD1000 Primary Flight Display (PFD)</td>
<td>CS 23</td>
<td>STC approved for Piper PA-24 series Carlisle, PA1788.</td>
<td>13/06/2009</td>
<td>Active</td>
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<tr>
<td>10026840</td>
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<td>TCCA LSTC C-L.5508-028/ D, ISS.1</td>
<td>Installation, Complete Custom Aircraft Interior.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCDA LSTC No. C- LSAS09-0280 Issue 1 dated 04 August 2009.Aero Consulting Services Ltd, Modification Data Summary G100000 Revision EMidcoast Aviation Inc/Aero Consulting Services Ltd Airplane Flight/Manu...</td>
<td>Restricted to ASN 8870 Prior to installation of this design change it must be determined that the interface between this design change and any other previously installed design change and/or repair will introduce a significant effect upon the aerodynamics...</td>
<td>16/06/2013</td>
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<tr>
<td>10026784</td>
<td></td>
<td></td>
<td>PA-46-500TP, PA-46R-350T, PA-46-310P</td>
<td>Installation of Single or Dual Garmin G500 / G500 Avionics System(INCLUDING SVS) and Dual Garmin GNS4330/XN GPS/COMM/NAV.</td>
<td>-EASA approved Master Document Ltd, Doc. No. ASR-2009-075- MDL-01-00,Amtd. Level 00, dated 20-Jun-2010, or later approved amdt. If EASA approved Instructions for Installation, Doc. No. ASR-2009-075- IFI-10-00, Amtd. Level 00, dated 18-Jun-2009, or later approved amdt. Level -EASA approved Flight Manual...</td>
<td>This STC is limited to the configuration of aircraft, equipment type and functions as listed in EASA approved Instructions for Installation Doc. No. ASR-2009-075-IFI-10-00, Amtd. Level 00, dated 18-Jun-2009, especially as mentioned in item 2. of this document; or later approved amendment...</td>
<td>23/04/2010</td>
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<td>10026785</td>
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<td>95-C55, D55, E55, 58, G58</td>
<td>Installation of 4-blade MTV-14-D-C-F-FC 188-30 propellers HawkerBeechcraft Model 95-C55, D55, E55, 58, G58.</td>
<td>FAR 23 including amendment 34</td>
<td>STC approved for Piper PA-24 series Carlisle, PA1788.</td>
<td>10/08/2009</td>
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<td>BEECH 1900D</td>
<td>Installation of Skytrac ISAT-100 GPS/Iridium communicator system</td>
<td>As defined in the FAA TCDS A24CE -Design Order Package DFE-03-0381 DOP-00, dated 15-05-09-Declaration of Compliance, dated 11-07-09</td>
<td>None.</td>
<td>11/06/2009</td>
<td>Active</td>
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<td>10026811</td>
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<td>BOEING 767-300</td>
<td>Installation of LIFEPdot Stretchers.</td>
<td>As per TCDS.</td>
<td>STC approved for Piper PA-24 series Carlisle, PA1788.</td>
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<td>PA-24, PA-24-250, PA-24-260, PA-24-400</td>
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<td>CS 23</td>
<td>STC approved for Piper PA-24 series Carlisle, PA1788.</td>
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<td>Active</td>
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<td>TCCA LSTC C-L.5508-028/ D, ISS.1</td>
<td>Installation, Complete Custom Aircraft Interior.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCDA LSTC No. C- LSAS09-0280 Issue 1 dated 04 August 2009.Aero Consulting Services Ltd, Modification Data Summary G100000 Revision EMidcoast Aviation Inc/Aero Consulting Services Ltd Airplane Flight/Manu...</td>
<td>Restricted to ASN 8870 Prior to installation of this design change it must be determined that the interface between this design change and any other previously installed design change and/or repair will introduce a significant effect upon the aerodynamics...</td>
<td>16/06/2013</td>
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List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<th>Date of Issue</th>
<th>Status</th>
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<tr>
<td>GOODRICH CORPORATION</td>
<td>ST008956E-D</td>
<td>Installation of the CHS (Cargo Handling System) on the B767-300FMD</td>
<td>The Certification Basis has been determined and is defined in CRI A-01. The Certification Basis for the original product shall remain unchanged and remain applicable to this certificate approval</td>
<td>FAA-CT-13582E-DOC-1 from 17 August 2018</td>
<td>The operation of the STC is only approved when installed on aircrafts with the Boeing A/C B767-300F.</td>
<td>15/08/2009</td>
<td>Active</td>
<td>10028859</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>CESSNA 500, 550</td>
<td>GARMIN GPS 400W/500W</td>
<td>Certification Basis as per EASA TCDS IM.A.207 Issue 2 from 5 March 2009</td>
<td>Master Document List &quot;2054628-DOO&quot; Issue 1 from 17 August 2009 Aircraft Flight Manual Supplement &quot;2054628-AM&quot; Issue 1 from 11 August 2009 Instructions for Continued Airworthiness &quot;2054628-CA&quot; Issue 1 from 11 August 2009 later revisions of the above listed documents approved by EASA</td>
<td>The limitations according to Aircraft Flight Manual Supplement &quot;2054628-AM&quot; are applicable.</td>
<td>18/08/2009</td>
<td>Active</td>
<td>10028864</td>
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<tr>
<td>DUNCAN AVIATION, INC.</td>
<td>CESSNA 650</td>
<td>Installation of an AlliedSignal Enhanced Ground Proximity Warning System (EGPWS)</td>
<td>The Certification Basis for the original product remains applicable to the above listed documents approved by EASA</td>
<td></td>
<td></td>
<td>18/08/2009</td>
<td>Active</td>
<td>10028867</td>
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<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>HAWKER 900 XP</td>
<td>Installation of a Honeywell Filipiak Engine Condition Trend Monitoring - Data Downloader (ECTM-DD)</td>
<td>FAR 25</td>
<td>Installation in accordance with Chipewa Aerospace Master Drawing List (MDL) No H800-7610-01-0001, Rev 11, dated February 28, 2009, or later approved revisions. Instructions for Continued Airworthiness in accordance with Chipewa Aerospace Instructions for Continued Airworthiness (ICA) No H800-7610-01-00...</td>
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<td>19/08/2009</td>
<td>Active</td>
<td>10028871</td>
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<td>HAWKER BEECHCRAFT SERVICES</td>
<td>HAWKER BEECHCRAFT B200GT, B200 BEECH B300C, BEECH B200CT, B200T, B300, BEECH 200, 300LW, B200, B200C, BEECH 200, 200C, 200CT, 200T</td>
<td>Installation of an AirLock Secure Entry Door Lock System in place of original equipment locks for the main cabin door, nose avionics doors, and aft equipment doors.</td>
<td>FAR 23</td>
<td>Installation in accordance with Hawker Beechcraft Services Master Data List Document NO 007-5200-L01-00 Revision 1 dated 5 March 2009, or later approved revisions.</td>
<td></td>
<td>13/08/2009</td>
<td>Active</td>
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<tr>
<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>CL-600-2D15, CL-600-2D2A, CL-600-2B10, CL-600-2C10, Goodrich Class III EFB System</td>
<td>The Certification Basis for the original product remains applicable to the above listed documents approved by EASA</td>
<td>- FAA STC No. ST03573AT dated 05 August 2010 - Chipewa Aerospace Master Data List (MDL) 601-4620-01-0001, Revision 4, dated 28 February 2009 - Chipewa Aerospace Airplane Flight Manual Supplement No. 601-4620-012B Revision 0R - Instructions for Continued Airworthiness (ICA), Chipewa Aerospace, Inc...</td>
<td>Restricted to Configuration 2 as defined in Chipewa Aerospace Certification Plan No 601-4620-CP Revision D dated 24 March 2009.</td>
<td>14/01/2013</td>
<td>Active</td>
<td>10028882</td>
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<td>EASA STC No.</td>
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<td>10026892</td>
<td>ROSEN SUNVISOR SYSTEMS L.L.C.</td>
<td>FAA STC SA26252NM</td>
<td></td>
<td>Installation of the Rosen sunvisor with a new CNSS-1610U Complete Flight Deck Credible System &amp; Installation Kit</td>
<td>Original EASA Certification Basis for FAA STC SA26252NM</td>
<td>None.</td>
<td>19/08/2009</td>
<td>Active</td>
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<tr>
<td>10026895</td>
<td>DONALDSON COMPANY, INC.</td>
<td>FAA STC SR02526CH</td>
<td>MD600</td>
<td>Installation of engine inlet barrier filter (IBF) systemRevision 1 reflects the name and address change from Aerospace/Intrusion Systems, Inc. to Donalsdon Company, Inc.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate; The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>1. Master Data List Inlet Barrier Filter system AFPS-MD600-IFC 2000 or later approved by EASA. 2. Aircraft Flight Manual Supplement AFPS-MD600-IFC/FMS REV A approved by the FAA on 21 July 2009.3. Instructions for Continued Airworthiness AFPS-MD600-IFC/CA-REV Dated 20 October 2006 or later revisions of the above listed documents.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>3/1/2018</td>
</tr>
<tr>
<td>10026896</td>
<td>OESSNA ZURICH CITATION SERVICE CENTUMBI</td>
<td>FAA STC SR02526CH</td>
<td>HS 125 SERIES 708</td>
<td>cockpit voice recorder installation by installing a new L3 Fr2100 cockpit voice recorder (CVR)</td>
<td>FAA CAR-4b, FAR 25 for the Change</td>
<td></td>
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<tr>
<td>10026904</td>
<td>ACS-NAI Ltd.</td>
<td>FAA STC SR02526CH</td>
<td>TC-842B19</td>
<td>Installation, Complete Custom Aircraft Interior.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>TC-842B19</td>
<td>Restricted to AS580. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/08/2013</td>
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<tr>
<td>10026905</td>
<td>JET AVIATION AG</td>
<td>FAA STC SR02526CH</td>
<td>737-700 IGW</td>
<td>Activation of Runway Awareness and Advisory System (RAAS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Jet Aviation Documents- Master Data List BB7373464-001 Rev B dated 31 July 2009- APM Supplement BB7373464-01-01 Rev B dated 20 October 2009.</td>
<td>Prior to installation of this modification the installation must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/04/2016</td>
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<tr>
<td>10026907</td>
<td>AIR COMM CORPORATION LLC</td>
<td>FAA STC SR02526CH</td>
<td>BELL 206</td>
<td>Bell Helicopter Model 407 Cabin Ventilation System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>FAR 27 D-407V revision 1 dated 16 April 2009 or later approved revision - Master Document Hel407V-1 revision 2 dated 19 May 2009 or later approved revision - Flight Manual Supplement 407V-2000M revision 1 dated 21 April 2009 or later approved revision - Instructions for Continued Airworthiness.</td>
<td>None</td>
<td>21/08/2009</td>
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<tr>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>FAA STC 10026912</td>
<td>FAA STC ST027403CH</td>
<td>Installation of improved parts for rotor transmission support in accordance with Carson Services Inc. Drawing Number 22000, Revision B dated 22 01 1994.</td>
<td>25/08/2009</td>
<td>Active</td>
<td>10026912</td>
</tr>
<tr>
<td>TROLLTUNE CORPORATION</td>
<td>FAA STC 10026913</td>
<td>182P AND 182Q (SEE ATTACHMENT)</td>
<td>Increased Maximum Gross Takeoff Weight (MGW/TW). This STC allows a 150 lbs increase in MGTOW from 2650 to 3100 lbs. There is no change to maximum landing weight of 2950 lbs.</td>
<td>25/08/2009</td>
<td>Active</td>
<td>10026913</td>
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<tr>
<td>RUAG AG</td>
<td>FAA STC 10026920</td>
<td>AS 332 C1</td>
<td>Pilot and Co-pilot Upper/Lower Cargo Hook Release modification Rev 1: Transfer of STC to a new legal entity that is holder of a new DOA(EASA,12.76)</td>
<td>03/07/2024</td>
<td>Active</td>
<td>10026920</td>
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<tr>
<td>AEROLITE AG</td>
<td>FAA STC 10026922</td>
<td>EC135 T2+</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>03/07/2024</td>
<td>Active</td>
<td>10026922</td>
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<tr>
<td>CARSON HELICOPTERS, INC.</td>
<td>FAA STC 10026925</td>
<td>S-61N</td>
<td>Installation of improved parts for rotor transmission support inaccordance with Carson Services Inc. Drawing Number 22000, Revision B dated 22 01 1994.</td>
<td>03/07/2024</td>
<td>Active</td>
<td>10026925</td>
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<tr>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>FAA STC 10026941</td>
<td>60, A60, B60</td>
<td>Installation must be in accordance with Flight Manual Supplement No. ASR2007-088- FMS-5.1-0 Amendment 00 dated 20-10-2008 or later approved revision.</td>
<td>03/07/2024</td>
<td>Active</td>
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<tr>
<td>ST006943</td>
<td>PATS AIRCRAFT L.L.C.</td>
<td>REV. 3</td>
<td>FAA ST01713NY</td>
<td>737-800</td>
<td>1. Installation of Auxiliary Fuel Systems in Boeing 737-800(BBJ2)aircraft. 25.981 amendment 25-125 Compliant. All configurations listed below are approved with or without Boeing Nitrogen Generation System (NGS) installed. 2. Configuration 1: Installation of a Four-Cell Auxiliary Fuel System on Forew...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification basis as per FAA Flight Manual Supplement 1, PATS Aircraft, LLC Airplane Flight Manual Supplement No. 132 for all Boeing 737-800 (BBJ2) aircraft. FAA approved revision dated February 6, 2006, or later FAA approved revision is required.</td>
<td>06/07/2021</td>
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<td>10020525</td>
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<td>STC No.</td>
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<td>10027068</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td></td>
<td>TC46 STC/SA04-113</td>
<td>BD-700-1A10</td>
<td>Installation, complete Custom Aircraft Interior</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval (CRI A-100, CRI D-100, CRI D-200, CRI D-300, CRI D-500, CRI F-100, CRI F-200) and/or emissions levels of the original.</td>
<td></td>
<td>Restricted to AS6936.1. Refer to the Limitations section of the required AFM Supplement and the referenced FCOM(US)2.2-825. The system is compatible only to IEE8802.11(b). Prior to installation of this modification the installer must determine that the interrelationship between this modification and...</td>
<td>02/05/2009</td>
<td>Active</td>
<td>10027068</td>
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<tr>
<td>10027064</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td>REV. 1</td>
<td>RALLYE 235 A, C, E, E-D, F, MS894 A, B, E, E-D</td>
<td>Reduction of maximum take off and landing weights to 1000 kg.</td>
<td>FAR 23</td>
<td>SEGA Flight Manual Supplement A-55-09-100kg, ed. 0, dtd. 20-07-2009, Ethernet PR version SEGA Installations Instructions A-55-09-100kg, ed. 0, dtd. 20-07-2009 SEGA Modification File A-55-09-100kg, ed. 0, dtd. 20-07-2009.</td>
<td>M5894 models: seating capacity limited to 3 persons, pilot included. Rallye 235 models: seating capacity limited to 3 persons, pilot included. Fuel tank capacity limited to 240 l (2&quot;120). The airplane configurations listed in attachment 1 are excluded from theapplicability of this STC because their...</td>
<td>08/01/2010</td>
<td>Active</td>
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<tr>
<td>10027066</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td></td>
<td>GV-SP (G550)</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5224.</td>
<td>JAR 25</td>
<td>Install in accordance with Gulfstream Aeriesp Report GCS20000634 Rev. N/C dated 11 June 2009 and Index List GCS20000647 Rev. 0, dated 06-08-09, or later approved revisions. Operate in accordance with Gulfstream Aeriesp Flight Manual Supplement GCS20000918 Rev. N/C dated June 10, 2009, or lat...</td>
<td>Aircraft Serial Number 5224 only.</td>
<td></td>
<td>03/06/2009</td>
<td>Active</td>
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<td>10027066</td>
<td>B/E AEROSPACE LIMITED</td>
<td>REV. 2</td>
<td>TC46 LSTC NO. C-LSA09-033/D</td>
<td>CL-860-2B19</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product ameunchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>Restricted to AS6936. Prior to installation of this designchange it must be determined thatthe interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>16/06/2013</td>
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0027110

LEARJET INC.

FAA STC
ST01716LA-D

Description
Installation of a Passenger Seat Upgrade: Validation of FAA STC ST01716LA-D

Certification basis
FAA STC ST01716LA-D Master Drawing List, ref. Doc. BAS88010001, Revision C, dated 28th February 2005- Maintenance Requirements - Instruction for Continued Airworthiness ref. doc. BAS88010001-PCA-1 dated 24th February 2005 (* or later revisions approved by EASA in accordance with EASA ED Decision)

None
07/09/2009
Active
10027110

0027112

DASSAULT FALCON JET CORP

FAA STC
SA8300SW-D

Description
Installation of the Automatic Cabin Electrical Load Shed System in accordance with Drawing List Report No. FEX-1011, Revision E dated 28th September 2000 or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/01CF (or subsequent revisions of this decision)

The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval

FAA STC SA8300SW-D Installation of the Automatic Cabin Electrical Load Shed System in accordance with Drawing List Report No. FEX-1011, Revision E dated 28th September 2000 or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/01CF (or subsequent revisions of this decision)

None
07/09/2009
Active
10027112

0027113

GAMA ENGINEERING LTD

CAA UK RESPONSIBIL ITY

PA-28 Aspen EFD1000 System

The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval

None
07/09/2009
Active
10027113

0027115

CESSNA DUESSLEDORF CITATIONSERVIC E CENTER GMBH

FAA STC
SA8283SW-D

Description
Installation of a TAWS System Honeywell Mark VII EGPA5

Certification Basis as per EASA TCDS M.A.S.080 Issue 2, 8th October 2008 UAR 25 Change 15 plus other airworthiness requirements, Special Conditions and Equivalent Safety Findings

FAA STC SA8283SW-D Installation of a TAWS System Honeywell Mark VII EGPA5 with Special Conditions and Equivalent Safety Findings

None
07/09/2009
Active
10027115

0027116

CESSNA DUESSLEDORF CITATIONSERVIC E CENTER GMBH

FAA STC
SA8300SW-D

Description
Installation of an FMS System Universal UNS-1M

Certification Basis as per EASA TCDS M.A.S.080 Issue 2, 8th October 2008 UAR 25 Change 15 plus other airworthiness requirements, Special Conditions and Equivalent Safety Findings

FAA STC SA8300SW-D Installation of an FMS System Universal UNS-1M with Special Conditions and Equivalent Safety Findings

None
07/09/2009
Active
10027116

0027121

AIRCRAFT PERFORMANCE MODIFICATIONS INC.

FAA STC
SA1905LA

Description
Installation of external aft body strakes, in accordance with FAA Approved Aircraft Performance Modifications Master Drawing List as listed in AML No. SA1905LA, Revision R, or later FAA approved revision

The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later installations of the original product or in accordance with FAA TCDS M.A.S.080 Issue 2, 8th October 2008 UAR 25 Change 15 and/or FAA approved revision.

FAA STC SA1905LA Installation of external aft body strakes, in accordance with FAA Approved Aircraft Performance Modifications Master Drawing List as listed in AML No. SA1905LA, Revision R, or later FAA approved revision

None
07/09/2009
Active
10027121
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<th>EASA STC No</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA C/T Code</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>10027122</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>REV 3</td>
<td>TCCA STO SA04-68</td>
<td>DHC-8-315, DHC-8-314, DHC-8-311, DHC-8-301, DHC-8-202, DHC-8-201, DHC-8-106, DHC-8-103, DHC-8-102</td>
<td>Retrofit installation of Universal Avionics Systems UNS-1F</td>
<td>Installation of Flight Management System.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Configured as EFIS-equipped Aircraft, Single UNS-1F</td>
<td>26/06/2019</td>
<td>Active</td>
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<td>10027141</td>
<td>DECA AVIATION ENGINEERING Ltd \ibia PAL AEROSPACE</td>
<td>REV 1</td>
<td>TCCA SA00-126; CAA NL SA0014NL</td>
<td>DHC-8-301, DHC-8-311, DHC-8-302, DHC-8-201, DHC-8-106, DHC-8-103, DHC-8-102</td>
<td>QAR/AAP Monitoring System/TCCA SA00-126, Iss. 4</td>
<td>FAR Part 25 as per TCDS EASA.IM.A.191 for DHC-8-100/200/300 Series</td>
<td>Approval limited to S/N RN073</td>
<td>01/09/2009</td>
<td>Active</td>
<td>10027141</td>
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<td>10027152</td>
<td>AVIATION SUPPORT GmbH</td>
<td></td>
<td>600N</td>
<td>Installation of HD Camera System and right side dual utility mount MD600N as per Master Document DECA-4903-02 Supplement DECA-4903-03</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Instruction for Continued Airworthiness Ref: ICAO-ASE-00819, Rev. 0.3 dated 08.09.2009. Flight Manual Supplement Ref: RMI-ASE-00819, Rev 0.4 dated 08.09.2009 and later revisions of the above listed documents approved by EASA.</td>
<td>Approval limited to S/N RN073</td>
<td>10/09/2009</td>
<td>Active</td>
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<td>10027156</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
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<td>FAA STC ST00644CH</td>
<td>L1011-385-1, L1011-385-1/4, L1011-385-1/5, L1011-385-3</td>
<td>Installation of a single or dual Honeywell Primus Prime P/N 9002 and associated hazardous loads.</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Approval limited to S/N RN073</td>
<td>10/09/2009</td>
<td>Active</td>
<td>10027156</td>
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<td>10027158</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>REV 1</td>
<td>CESSNA 525</td>
<td>Honeywell GNS-XLS P/NAV Upgrade</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>-2008-4055, Rev. N/C, Airplane Flight Manual Supplement Revision 10</td>
<td>Restricted to aircraft with S/N 0001 to 0099 which already have an EFIS/Honeywell GNS-XLS installed. Prior to installation of this design change it must be determined that the interrelationships between this design change and any other previous/installed design change and/ or repair will introduce no adverse...</td>
<td>29/01/2014</td>
<td>Active</td>
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<td>10027192</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
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<td>BD-700-1A10, -1A11</td>
<td>EGPSW MK V Enhancement with RAAS Upgrade of the existing Honeywell Enhanced Ground Proximity Warning System (EGPSW) MK V with the Honeywell Runway Awareness and Advisory System (RAAS). Installation of an optional RAAS Advisory Mute switch into the center pedestal.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List, 29IC-ML Revision A dated 10 September 2009 - Approval for Aircraft Modification, 29IC-CN Revision A dated 10/N/2009 and later revisions of the above listed documents approved by EASA.</td>
<td>- BD-700-1A10: S/N 0002 and subsequent, BD-700-1A11: S/N 9127 and subsequent</td>
<td>14/09/2009</td>
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<td>ROSEN SUNVISOR SYSTEMS L.L.C.</td>
<td>10027193</td>
<td>FAA STC SA2614NM</td>
<td>2</td>
<td>MYSTERE FALCON 50 1, MYSTERE FALCON 200</td>
<td>EASA Validation of FAA STC SA2614NM/coolSpot Sun Visor Installation in accordance with Rosen Drawing List No. RF 50/2000-000L, Revision C dated 28th August 2009 or later revisions/approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>The Certification Basis for the original product remains applicable to all FAA STC SA2614NM/coolSpot Sun Visor Installation installations</td>
<td>FAA STC SA2614NM/coolSpot Sun Visor Installation Installation of FAA STC SA2614NM/coolSpot Sun Visor Installation in accordance with Rosen Drawing List No. RF 50/2000-000L, Revision C dated 28th August 2009 or later revisions/approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>None</td>
<td>15/09/2009</td>
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<td>TIMCO AVIATION SERVICES INC.</td>
<td>10027196</td>
<td>FAA STC ST00596AT-D</td>
<td>8</td>
<td>ALPHA AVANTI II Installation of installation of Satcom System with Ball Antenna's in accordance with Timco Master Data List MDL-01088, Revision M, dated October 27, 1999 or later approved revision.</td>
<td>Installation of Collins SAT-906 SatCom System with Ball Antenna's in accordance with Timco Master Data List MDL-01088, Revision M, dated October 27, 1999 or later approved revision.</td>
<td>The Certification Basis for the original product remains applicable to all FAA STC ST00596AT-D installations</td>
<td>FAA STC ST00596AT-D</td>
<td>None</td>
<td>14/09/2009</td>
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<tr>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER Gmbh</td>
<td>10027199</td>
<td>P.180 AVANTI II</td>
<td>1</td>
<td>Installation of Garmin GPS-400 / 400W System</td>
<td>Installation of Garmin GPS-400 / 400W System</td>
<td>The Certification Basis for the original product remains applicable to all Garmin GPS-400 / 400W System installations</td>
<td>Garmin GPS-400 / 400W System</td>
<td>None</td>
<td>14/09/2009</td>
<td>Active</td>
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<td>RUAG AG</td>
<td>10027200</td>
<td>REV. 1</td>
<td>MD900</td>
<td>TV Broadcasting Equipment installation (camera adaptation on existing mount and certification of the operators manual)</td>
<td>TV Broadcasting Equipment Installation (camera adaptation on existing mount and certification of the operators manual)</td>
<td>The Certification Basis for the original product remains applicable to all TV Broadcasting Equipment installations</td>
<td>TV Broadcasting Equipment Installation (camera adaptation on existing mount and certification of the operators manual)</td>
<td>None</td>
<td>18/01/2021</td>
<td>Active</td>
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<td>PREMIAR AIRCRAFT ENGINEERING LTD</td>
<td>10027205</td>
<td>FAA STC SA1873SE</td>
<td>7</td>
<td>Beech 390 (Premier 1)</td>
<td>Beech 390 (Premier 1)</td>
<td>The Certification Basis for the original product remains applicable to all Beech 390 (Premier 1) installations</td>
<td>Beech 390 (Premier 1)</td>
<td>None</td>
<td>14/09/2009</td>
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<td>Limitations/Conditions</td>
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<td>10027217</td>
<td>AIR ALLIANCE GmbH</td>
<td>REV. 1</td>
<td>T8A, T8B10, TB 20, TB21, TB200</td>
<td>Installation of Aspen EFD 1000 PRO</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/ approval. CERTIFICATION Document No. 10027217 dated 12/04/2010 none.</td>
<td>None</td>
<td>12/04/2010</td>
<td>Active</td>
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<td>10027220</td>
<td>VIDEO TECHNOLOGY SERVICES, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST0725NY</td>
<td>Installation of Digital Video Player (DVP)</td>
<td>Certification Basis in accordance with EASA TCDS IM.A.035 plus CRI 6-11 Special Condition H-01 &quot;Enhanced Airworthiness Programme for Airplane Systems - ICAS&quot; and EASA Certificate of Airworthiness. CERTIFICATION Document No. 10027220 dated 03/07/2024 none.</td>
<td>None</td>
<td>03/07/2024</td>
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<td>10027223</td>
<td>PATS AIRCRAFT SYSTEMS d/b/a ALOFT AEROCRITECT S</td>
<td>REV. 2</td>
<td>FAA STC ST0173NY-D</td>
<td>Installation of an Executive Interior</td>
<td>EASA STC No. 10027223 Revision 1 dated 08 February 2010 FAA STC No. ST0173NY-D dated 14 January 2011 and PATS Aircraft, LLC Master Drawing List 6049-001- MDL, revision F dated 01 July 2006 PATS Aircraft, LLC. Aircraft Flight Manual Supplement No. 212 for an Executive Interior FAA approved.</td>
<td>In this installation the maximum passenger seating capacity is limited to 15.</td>
<td>12/07/2012</td>
<td>Active</td>
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<td>10027252</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>REV. 1</td>
<td>TCCA STC SA09-15</td>
<td>Iceland's Coast Guard Maritime Patrol Modification TCCA STC Number: SA09-15</td>
<td>Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/ approval. CERTIFICATION Document No. 10027252 dated 16/09/2009 none.</td>
<td>U.S.-CFA Flight Management Systems approved for EN-route, Terminal and Non Precision approach operations. Only External LAN System not approved for airborne operations.</td>
<td>16/09/2009</td>
<td>Active</td>
<td>10027252</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10027287</td>
<td>REV. 3</td>
<td>FAA STC</td>
<td>ST02756CH-D</td>
<td>Installation of ECS EZMount Mechanical Mounting Assemblies for the original product as amended by the following additional or alternative airworthiness requirements: The applicant is elected to comply to the following paragraphs at C25 Amendment 4: Configuration 1: EZMount® Slider 737-2, EZMount® Pivot 737-3, EZMount® Slider 737-5, EZMount® Pivot 737-7, EZMount® Slider 737-8, Revision N/C, and any other previously installed change/repair.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The applicant is elected to comply to the following paragraphs at C25 Amendment 4: Configuration 1: EZMount® Slider 737-2, EZMount® Pivot 737-3, EZMount® Slider 737-5, EZMount® Pivot 737-7, EZMount® Slider 737-8, Revision N/C, and any other previously installed change/repair.</td>
<td>Standard Aero Documents: Master Data List MDL1104-100 Rev. B dated September 09, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA certification basis as per EASA TCDS IM.A.080 Issue 2, 8th October 2008 (JAR 25 Change 15 plus other airworthiness requirements, Special Conditions and Equivalent Safety Findings).</td>
<td>04/06/2014</td>
<td>Active</td>
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<td>10027284</td>
<td>REV. 3</td>
<td>FAA STC</td>
<td>ST02837CH</td>
<td>Installation of ECS EZMount Mechanical Mounting Assemblies for the original product as amended by the following additional or alternative airworthiness requirements: The applicant is elected to comply to the following paragraphs at C25 Amendment 4: Configuration 1: EZMount® Slider 737-2, EZMount® Pivot 737-3, EZMount® Slider 737-5, EZMount® Pivot 737-7, EZMount® Slider 737-8, Revision N/C, and any other previously installed change/repair.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The applicant is elected to comply to the following paragraphs at C25 Amendment 4: Configuration 1: EZMount® Slider 737-2, EZMount® Pivot 737-3, EZMount® Slider 737-5, EZMount® Pivot 737-7, EZMount® Slider 737-8, Revision N/C, and any other previously installed change/repair.</td>
<td>Standard Aero Documents: Master Data List MDL1104-100 Rev. B dated September 09, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA certification basis as per EASA TCDS IM.A.080 Issue 2, 8th October 2008 (JAR 25 Change 15 plus other airworthiness requirements, Special Conditions and Equivalent Safety Findings).</td>
<td>09/11/2021</td>
<td>Active</td>
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<td>FAA STC</td>
<td>ST02837CH</td>
<td>Installation of ECS EZMount Mechanical Mounting Assemblies for the original product as amended by the following additional or alternative airworthiness requirements: The applicant is elected to comply to the following paragraphs at C25 Amendment 4: Configuration 1: EZMount® Slider 737-2, EZMount® Pivot 737-3, EZMount® Slider 737-5, EZMount® Pivot 737-7, EZMount® Slider 737-8, Revision N/C, and any other previously installed change/repair.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The applicant is elected to comply to the following paragraphs at C25 Amendment 4: Configuration 1: EZMount® Slider 737-2, EZMount® Pivot 737-3, EZMount® Slider 737-5, EZMount® Pivot 737-7, EZMount® Slider 737-8, Revision N/C, and any other previously installed change/repair.</td>
<td>Standard Aero Documents: Master Data List MDL1104-100 Rev. B dated September 09, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA certification basis as per EASA TCDS IM.A.080 Issue 2, 8th October 2008 (JAR 25 Change 15 plus other airworthiness requirements, Special Conditions and Equivalent Safety Findings).</td>
<td>04/06/2014</td>
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<td>FAA STC</td>
<td>ST02837CH</td>
<td>Installation of ECS EZMount Mechanical Mounting Assemblies for the original product as amended by the following additional or alternative airworthiness requirements: The applicant is elected to comply to the following paragraphs at C25 Amendment 4: Configuration 1: EZMount® Slider 737-2, EZMount® Pivot 737-3, EZMount® Slider 737-5, EZMount® Pivot 737-7, EZMount® Slider 737-8, Revision N/C, and any other previously installed change/repair.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The applicant is elected to comply to the following paragraphs at C25 Amendment 4: Configuration 1: EZMount® Slider 737-2, EZMount® Pivot 737-3, EZMount® Slider 737-5, EZMount® Pivot 737-7, EZMount® Slider 737-8, Revision N/C, and any other previously installed change/repair.</td>
<td>Standard Aero Documents: Master Data List MDL1104-100 Rev. B dated September 09, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA certification basis as per EASA TCDS IM.A.080 Issue 2, 8th October 2008 (JAR 25 Change 15 plus other airworthiness requirements, Special Conditions and Equivalent Safety Findings).</td>
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<td>10027294</td>
<td>REV. 4</td>
<td>FAA STC</td>
<td>ST02837CH</td>
<td>Installation of ECS EZMount Mechanical Mounting Assemblies for the original product as amended by the following additional or alternative airworthiness requirements: The applicant is elected to comply to the following paragraphs at C25 Amendment 4: Configuration 1: EZMount® Slider 737-2, EZMount® Pivot 737-3, EZMount® Slider 737-5, EZMount® Pivot 737-7, EZMount® Slider 737-8, Revision N/C, and any other previously installed change/repair.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The applicant is elected to comply to the following paragraphs at C25 Amendment 4: Configuration 1: EZMount® Slider 737-2, EZMount® Pivot 737-3, EZMount® Slider 737-5, EZMount® Pivot 737-7, EZMount® Slider 737-8, Revision N/C, and any other previously installed change/repair.</td>
<td>Standard Aero Documents: Master Data List MDL1104-100 Rev. B dated September 09, 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA certification basis as per EASA TCDS IM.A.080 Issue 2, 8th October 2008 (JAR 25 Change 15 plus other airworthiness requirements, Special Conditions and Equivalent Safety Findings).</td>
<td>04/06/2014</td>
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<td>STC Holder</td>
<td>REV. 2</td>
<td>Foreign/NA</td>
<td>STC ref.</td>
<td>Model</td>
<td>Cabin Recomfiguration from Dragon Aviation Leasing to Juneyao Standard (DHL) Revision of the EASA STC 10027295 to remove EWSI limitation in accordance with the EASA TCDST. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable.</td>
<td>Certification Program, ref. doc. No. CP-A320-09-009. Revision 0 dated 31st August 2009- Statement of Compliance, ref. doc. No. SDCP-A320-09-009-005, dated 31st August 2009- Instruction for Continued Airworthiness, ref. doc. No. ICA-A320-09-009Rev. 02, dated 11th August 2010- Enhanced Zonal Analyses.</td>
<td>None</td>
<td>23/11/2010</td>
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<td>10027297</td>
<td>FAA STC STO1727NY-D</td>
<td>REV. 4</td>
<td>Foreign/NA</td>
<td>ERU 190-100 ECJ</td>
<td>Model</td>
<td>Installation of an Executive Interior Major Change to FAA STC STO1727NY-D: Installation of an Executive Interior Major Change to FAA STC STO1727NY-D: Compliance with EWS ICA Special Condition H-01</td>
<td>Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CRI A-1 IMA S. 03026 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain...</td>
<td>Configuration 1: Master Drawing List 9590-0000-MDL Rev. D dated 16 April 2009 (&quot;Airplane Flight Manual Supplement No. 215, Rev. D dated 12 March 2016&quot;) (&quot;Maintenance Manual Supplement TO-3163 Rev. IR dated 1 May 2006&quot;) with TR20-1001 dated 10 April 2009.</td>
<td>24/06/2013</td>
<td>Active</td>
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<td>10027298</td>
<td>FAA STC STO1728NY-D</td>
<td>REV. 2</td>
<td>Foreign/NA</td>
<td>ERU 190-100 ECJ</td>
<td>Model</td>
<td>Modification of the Class C Forward Cargo Compartment to accommodate a new Class C Cargo Compartment: The Rear of the Main Cabin Major Change to FAA STC STO1728NY-D: Installation of an General Seavus Oxygen System: Major Change to FAA STC STO1728NY-D: Compliance with EWS ICA Special Condition H-01</td>
<td>Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CRI A-1 IMA S. 03026 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain...</td>
<td>Configuration 1: Modification of the Class C Forward Cargo Compartment to accommodate an Auxiliary Fuel Tank and Installation of a new Class C Cargo Compartment: The Rear of the Main Cabin: Master Drawing List 6671-0000-MDL Rev. IR dated 7 October 2008 (&quot;Airplane Flight Manual Supplement No. 211, R...&quot;).</td>
<td>24/06/2013</td>
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<td>10027299</td>
<td>FAA STC STO1729NY-D</td>
<td>REV. 2</td>
<td>Foreign/NA</td>
<td>ERU 190-100 ECJ</td>
<td>Model</td>
<td>Installation of a Passenger Gaseous Oxygen Supply System: Major Change to FAA STC STO1729NY-D: Compliance with EWS ICA Special Condition H-01</td>
<td>Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CRI A-1 IMA S. 03026 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain...</td>
<td>Master Drawing List 6672-0000-MDL Rev. A dated 15 December 2008 (&quot;Airplane Flight Manual Supplement No. 226, Rev. IR dated 17 April 2008&quot;). Maintenance Manual Supplement TO-3163 Rev. IR dated 1 May 2008 (&quot;...&quot; with TR20-1001 dated 10 April 2009&quot;) or later revisions of the above listed documents approved...</td>
<td>24/06/2013</td>
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<td>10027300</td>
<td>FAA STC SAO1597SE</td>
<td>REV. 2</td>
<td>Foreign/NA</td>
<td>LC42-550FG, T240 LC40-550FG, LC41-550FG</td>
<td>Model</td>
<td>Installation of replacement seats</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate approval.</td>
<td>According to DMS Flight Manual Supplement (IFMS) Date: N.M. SUP 5, revised dated 08/08/2009.</td>
<td>20/01/2017</td>
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<td>10027304</td>
<td>FOCA STC Z25-70-09</td>
<td>REV. 2</td>
<td>Foreign/NA</td>
<td>AS 305B, -81, -82, -83</td>
<td>Model</td>
<td>Installation of a new Class C Cargo Compartment: The Rear of the Main Cabin Major Change to FAA STC Z25-70-09: Compliance with EWS ICA Special Condition H-01</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CRI A-1 IMA S. 03026 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain...</td>
<td>FAAR 27 amendment 10 as defined in TCDST EASA R.008.</td>
<td>18/09/2009</td>
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<td>BOMBARDIER</td>
<td>ST027330</td>
<td>FAA STC ST03710AT-D</td>
<td>Installation of an Airplane Interior and completion for Aircraft Serial Number 4161</td>
<td>JAR 25. The Certification Basis for the original product remains applicable to this certificate.</td>
<td>- Master Drawing List ref. EASA-STC-1157A, Rev NC dated 31 August 2009 - Continued Airworthiness Instructions (CAI) ST03710AT-D - Installation Guide - Master Data List - Installation Guide for Aircraft Serial Number 4161</td>
<td>Aircraft Serial Number 4161 only. The following STCs are a required part of this STC: ST02987AT-D, ST03290AT-D, ST02979AT-D, ST03410AT-D</td>
<td>21/09/2009</td>
<td>Active</td>
<td>ST027330</td>
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<tr>
<td>BUSINESS AND COMPUTER AIRCRAFT</td>
<td>ST027323</td>
<td>FAA STC ST03710AT-D</td>
<td>Installation of an Airplane Interior and completion for Aircraft Serial Number 4161</td>
<td>JAR 25. The Certification Basis for the original product remains applicable to this certificate.</td>
<td>- Master Drawing List ref. EASA-STC-1157A, Rev NC dated 31 August 2009 - Continued Airworthiness Instructions (CAI) ST03710AT-D - Installation Guide - Master Data List - Installation Guide for Aircraft Serial Number 4161</td>
<td>Aircraft Serial Number 4161 only. The following STCs are a required part of this STC: ST02987AT-D, ST03290AT-D, ST02979AT-D, ST03410AT-D</td>
<td>21/09/2009</td>
<td>Active</td>
<td>ST027323</td>
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<tr>
<td>E.A.D AEROSPACE AIRWORTHINESS</td>
<td>ST027343</td>
<td>FAA STC ST03710AT-D</td>
<td>Installation of an Airplane Interior and completion for Aircraft Serial Number 4161</td>
<td>JAR 25. The Certification Basis for the original product remains applicable to this certificate.</td>
<td>- Master Drawing List ref. EASA-STC-1157A, Rev NC dated 31 August 2009 - Continued Airworthiness Instructions (CAI) ST03710AT-D - Installation Guide - Master Data List - Installation Guide for Aircraft Serial Number 4161</td>
<td>Aircraft Serial Number 4161 only. The following STCs are a required part of this STC: ST02987AT-D, ST03290AT-D, ST02979AT-D, ST03410AT-D</td>
<td>21/09/2009</td>
<td>Active</td>
<td>ST027343</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>ST027345</td>
<td>FAA STC ST03710AT-D</td>
<td>Installation of an Airplane Interior and completion for Aircraft Serial Number 4161</td>
<td>JAR 25. The Certification Basis for the original product remains applicable to this certificate.</td>
<td>- Master Drawing List ref. EASA-STC-1157A, Rev NC dated 31 August 2009 - Continued Airworthiness Instructions (CAI) ST03710AT-D - Installation Guide - Master Data List - Installation Guide for Aircraft Serial Number 4161</td>
<td>Aircraft Serial Number 4161 only. The following STCs are a required part of this STC: ST02987AT-D, ST03290AT-D, ST02979AT-D, ST03410AT-D</td>
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<td>Limitations/Conditions</td>
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<tr>
<td>10027347</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV 2</td>
<td>SA226T</td>
<td>Installation of MIC2X Camera system Installation of Wescam MX series (10, 10, 20) Gimbal Cameras with manual rotation / extension mechanism, operator console and downlink antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Document List MDOCL_19719_rev1_2013 0501- AIRM Supplement SPON_19719_rev1_2013 0501- Installation instruction ii. 19719_rev1_20130501 or later revisions of the above listed documents approved by EASA.</td>
<td>According limitation section of AFM. Aircraft must be previously modified with camera hatches i.e. Swearingen Camera Hole Modification. Prior to installation of this design change it must be determined that the interface between the design change and any other previously installed design change...</td>
<td>24/09/2009</td>
<td>Active</td>
<td>10027347</td>
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<tr>
<td>10027351</td>
<td>DASSAULT FALCON SERVICE</td>
<td>-</td>
<td>-</td>
<td>P-RNAV Capability</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Compliance has been shown in accordance with the requirements laid down by JAA Temporary Guidance Leaflet n. 10 “Airworthiness and Operations…”</td>
<td>- DFS Change Approval Sheet DFS 5559-CAS Issue A, dated 21st September 2009...DFS Aeroplane Flight Manual Supplement Number 48, Issue 2, dated 21st September 2009</td>
<td>This EASA approval is applicable on those Mystere Falcon 50 aircraft which have previously incorporated the Mod DFS 5559 approved under EASA.A.S.03915 with...Collins Autopilot APS 85-DFS B6 C14 and...Version 802.X for the FMS UNS-TL, Software. This design change is not applicable on the F50EX aircraft...</td>
<td></td>
<td>24/09/2009</td>
<td>Active</td>
<td>10027351</td>
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<tr>
<td>10027352</td>
<td>BOMBARDIER INC.</td>
<td>-</td>
<td>TCCA-STC SA88-90</td>
<td>Installation of Cockpit Speaker Mute Switch</td>
<td>EASA TCDS No. IM.A.0097 The Certification Basis for the original product remains applicable to this certificate. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>TCCA STC No. SA88-90 Issue 1 dated 18 December 2008 TCACSCo reference C-09-0252 dated 17 September 2009 Aero Consulting Services Modification Data Summary Number 007600 Revision NIC Aero Consulting Services Instructions for Continued Airworthiness 207600- ICA or later revisions of the above listed...</td>
<td>None.</td>
<td></td>
<td>24/09/2009</td>
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## List of EASA Supplemental Type Certificates

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<th>Certification basis</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY/NTER USA, INC.</td>
<td>Installation of a Honeywell HF Communications System, Model HRF-1050 and additionally the Avtech Corporation SELCAL decoder, Model CSD-714.This STC was originally issued as a validation of Brazilian CTS (STC/MAA, 2009/05-14 (20 May 2009), and then transferred to a United States based Design Approval Holder (D...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce or exacerbate any effect upon the airworthiness of the product.</td>
<td>26/11/2019</td>
<td>Active</td>
<td>10027372</td>
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<tr>
<td>DANSK FLY ELEKTRONIK ApS</td>
<td>Installation of ASPEN EF100 Pro Flight display</td>
<td>CS 23 Amendment 1</td>
<td>None</td>
<td>25/09/2009</td>
<td>Active</td>
<td>10027379</td>
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<tr>
<td>CESINA DUSELZDORF CITATIONSERVIC E CENTER GmbH</td>
<td>Installation of Honeywell KLJ-94-GPS</td>
<td>EASA TCDS IM.A.207The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The installation is limited to Cessna models 500, 550 and 560 equipped with Honeywell SPZ 550 Auto Pilot Flight Director System and RD44 Horizontal Situation Indicator System. Limitations according to AFMS 2008-8-8</td>
<td>28/09/2009</td>
<td>Active</td>
<td>10027391</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>FAA STC SA8279W-D Installation of the Auxiliary Battery System in accordance with Drawing List Report No. F9EX-01008, Revision K dated 26th September 2002 and all revisions approved by EASA in accordance with EASA ED Decision 2004/04/01 (or subsequent revisions of this decision).</td>
<td>The Certification Basis for the original product remains applicable. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>This design change is also applicable on the Falcon 900C designation as defined on EASA TCDS A.062</td>
<td>28/09/2009</td>
<td>Active</td>
<td>10027395</td>
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<tr>
<td>BOMBARDIER INC.</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>EASA STC No. SA04-113</td>
<td>Transport Canada SoC reference C-09-0644 dated 15 September 2009 TCCA STC No. SAA-113 Issue 1 dated 28 October 2004. Aeronautical Services Configuration Definition List D850409/Revision DL dated 08 September 2009. Instru...</td>
<td>Restricted to ASN 9303 Refer to the Limitations section(s) of the TCCA Designation. No reference to the FCOM special. Refer to the referenced system. Only compatible to IEEE 802.11 (b).</td>
<td>28/09/2009</td>
<td>Active</td>
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<td>EASA STC no.</td>
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<td>10027406</td>
<td>NSE</td>
<td>REV. 6</td>
<td>E 120 B</td>
<td>EC 120 B</td>
<td>Installation of NSE UMS Pin REC-4AR-O- TT-00Rev. 1. Software change in the Data Monitoring System “Brite Saver” 2 and an RFID Supplement added Rev. 2: Usage Monitoring System for EC120 – “Brite Saver” 2, Art. 4 added Rev. 3. Technical documentation revision numbers or levels Rev. 4: STC holder change from...</td>
<td>The Certification Basis for the original product as amended by (following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS-27 Annex 4 (selected to comply). CS 27.1316, CS 27.1317. The requirements for environmental protection and the associated...</td>
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<td>10027409</td>
<td>AERODATA AG</td>
<td>B300</td>
<td>Installation of Aerodata Flight Inspection System AD-AFS-150</td>
<td>The following new Certification Basis is applicable to this change (“CS-23, Annex 1, is elected to comply with for reasons affected by the change” The certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval</td>
<td>EASA Certification Basis as per EASA TCDS-R 008 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval The change comprises CS 27 First Issue: The certificate noise and/or emissions levels of the product...</td>
<td>-Dossier d'approbation de modification N-EC130 AS350B3 Approbation -F-SA-02, dated 23.07.2009- Dossier de Définition de la Modification N-EC130 AS350B3 Definition F- A-03, dated 26.06.2009 Dossier d'Installation N-EC130 AS350B3 Installation F- A-04, dated 28.01.2020 or later revisions of the above listed docs...</td>
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<td>10027424</td>
<td>TELEDYNE CONTROLS - FLIGHT DATACOMPANY</td>
<td>MD-80 SERIES</td>
<td>Installation of Wireless Quick Access Recorder.</td>
<td>FAR 25, Master Drawing List MOD- MDL-022-02 Issue 1 dated 30 June 2000.</td>
<td>-Installation of Safetyplane Equipment - Installation of a system for tracking a set of flight and system parameters during operation of the rotorcraft. The modification is based on change N-EC130 AS350B3 as defined in Dossier de Définition de la Modification N-EC130 AS350B3, dated 26.06.2009. EASA Certification Basis as per EASA TCDS-R 008 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval The change comprises CS 27 First Issue: The certificate noise and/or emissions levels of the product...</td>
<td>-Dossier d'approbation de modification N-EC130 AS350B3 Approbation -F-SA-02, dated 23.07.2009- Dossier de Définition de la Modification N-EC130 AS350B3 Definition F- A-03, dated 26.06.2009 Dossier d'Installation N-EC130 AS350B3 Installation F- A-04, dated 28.01.2020 or later revisions of the above listed docs...</td>
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<tr>
<td>10027431</td>
<td>ISEI</td>
<td>AS350 B3, EC130 B4</td>
<td>-Installation of Safetyplane Equipment - Installation of a system for tracking a set of flight and system parameters during operation of the rotorcraft. The modification is based on change N-EC130 AS350B3 as defined in Dossier de Définition de la Modification N-EC130 AS350B3, dated 26.06.2009. EASA Certification Basis as per EASA TCDS-R 008 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval The change comprises CS 27 First Issue: The certificate noise and/or emissions levels of the product...</td>
<td>-Dossier d'approbation de modification N-EC130 AS350B3 Approbation -F-SA-02, dated 23.07.2009- Dossier de Définition de la Modification N-EC130 AS350B3 Definition F- A-03, dated 26.06.2009 Dossier d'Installation N-EC130 AS350B3 Installation F- A-04, dated 28.01.2020 or later revisions of the above listed docs...</td>
<td>None</td>
<td>30/09/2009</td>
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<td>10027439</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>BAE JETSTREAM 3100, 3200</td>
<td>Installation of 5-Blade MTV-27-1-E-C-F-R (GJ/CFRL250-103) propellers</td>
<td>FAR 25, Annex 1 through 34 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval This certificate/approval involves a change to the certificate noise and/or emissions levels of the product...</td>
<td>-According to MT-Propeller Master Document List Doc. No. E-1764, Rev 1 and MT-Propeller Installation Instructions Doc. No. E-1766, Rev. 1, to be operated in accordance with MT-Propeller AFM/SM Document No E-1765, Original Issue, or later approved Revisions. To be maintained i.a.w. MT-Propeller ICA Doc...</td>
<td>None</td>
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<td>10027446</td>
<td>BRISTOW HELICOPTERS LTD</td>
<td>AS 332 L1</td>
<td>ECT Night Vision Imaging System InstallationMaster Drawing NOR-0450 Rev. Efbtr. STC revised due to new ownership Bristow Helicopters</td>
<td>As defined in TCDS EASA R. 002 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this...</td>
<td>Flight Manual Supplement AS332L-PFM- SUPP-0024 or later revisions of the above listed documents approved by EASA Master Drawing NOR-0450 Rev. E Minimum Equipment List Supplement AS332L-MEL-SUPP-002 Rev. E Minimum Equipment List Supplement AS332L-MEL-SUPP-002...</td>
<td>None</td>
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<tr>
<td>RUECKER AEROSPACE GmbH</td>
<td>208B</td>
<td>Installation of UHF Research Mission Equipment</td>
<td>DORNIER 228-101</td>
<td>The following new Certification Basis is applicable to the aircraft.</td>
<td>EASA approved Master Document, Doc. No. MLD-890-00021-000, Issue D, or later approved issue</td>
<td>Limited to serial number 7083</td>
<td>01/10/2009</td>
<td>Active</td>
<td>10027441</td>
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<tr>
<td>RUECKER AEROSPACE GmbH</td>
<td>FAA ST01850LA</td>
<td>Installation of UHF Research Mission Equipment</td>
<td>208B</td>
<td>The following new Certification Basis is applicable to the aircraft.</td>
<td>EASA approved Master Document, Doc. No. MLD-890-00021-000, Issue E, or later approved issue</td>
<td>Limited to serial number 7083</td>
<td>01/10/2009</td>
<td>Active</td>
<td>10027451</td>
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<td>SECURAPLANE TECHNOLOGIES, Inc.</td>
<td>FAA ST01850LA</td>
<td>SEE APPROVED MODEL LIST (AML)</td>
<td>FAR 25</td>
<td>Installation in accordance with Approved Master Data List (MDL) Document No. 170-04G0-01 Revision 1, dated April 13, 2005, or later approved revisions.</td>
<td>This approval is limited to those aircraft listed in the AML and equipped with lead acid battery. Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effects.</td>
<td>27/11/2014</td>
<td>Active</td>
<td>10027470</td>
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<td>H4 AEROSPACE (UK) Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B737-800</td>
<td>Introduction of Recaro Economy Passenger Seats on Boeing 737-800, in accordance with project H4A-1239. (Revision of STC 10027495 to remove EIS G Limitation)</td>
<td>Modification Summary Sheet: H4A1239, Issue 1 signed 02 October 2009, Instructions for Continued Airworthiness: H4A1239 Issue 2.1, dated April 13, 2005, or later approved revisions.</td>
<td>None</td>
<td>06/03/2011</td>
<td>Active</td>
<td>10027495</td>
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<td>LUFTHANSA TECHNICK AG</td>
<td>REV. 1</td>
<td>FOKKER F27 MARK 050</td>
<td>F700-EB3-034, Revision 01</td>
<td>Installation of self-illuminating Floor Path Marking System.</td>
<td>No. AFMS-890-00001-000A, Issue E, or later approved issue</td>
<td>None</td>
<td>06/10/2009</td>
<td>Active</td>
<td>10027505</td>
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<tr>
<td>SAGEM AVIONICS, LLC,</td>
<td>REV. 3</td>
<td>FAA STC SR09450RC</td>
<td>EC120B</td>
<td>SAGEM Integrated Cockpit Display System ICS-8A Installation Initial STC: Installation of SAGEM ICS-8A Integrated Cockpit Display System (ICDS) coupled to SAGEM autopilot per FAA STC SR09450RC-D (EASA IM R S. 01056) in accordance with SAGEM MDL 120-0002, Rev. D, and 1. Configuration per SAGEM drawing 2.</td>
<td>Rotorcraft Flight Supplement for Airbus Helicopters Model EC120 Baseline equipped with SAGEM ICS-8A Integrated Cockpit Display System (Software/Version EC120-AP95-4.8), SA-ER9-00722-01, Rev 00, dated 8 December 2017 for later revisions of the above listed documents approved by EASA.</td>
<td>For Night VFR operations the helicopter must be fitted with the back-up instrument permitting attitude control, such as a stand-by attitude indicator or a slip/skid indicator. Prior to installation of this design change it must be determined that the relationship between this design change and any of...</td>
<td>18/01/2017</td>
<td>Active</td>
<td>10027505</td>
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<td>STC Holder</td>
<td>System</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>FAA STC ST02761CH</td>
<td>Installation of Rosemount ST7002C-7 Series Laptop Docking Station (LDS) - Electronic Flight Bag (EFB) System. Validation of FAA STC ST02761CH</td>
<td>The Certification Basis for the original product and the following: Additional or alternative airworthiness requirements are applicable to the certificated aircraft. Refer to list of changes on FAA ST027611CH. The certificated noise and/or emissions levels of the original product are unchanged and remain...</td>
<td>Rosemount Aerospace Documents Master Data List D0403164 Rev. C (approved by the FAA)</td>
<td>Refer to limitations and conditions on FAA ST02761CH.</td>
<td>09/10/2009</td>
<td>Active</td>
<td>ST027532</td>
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<td>AEROSPACE DESIGN AND ENGINEERINGCO NSULTANTS Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>Traffic Alert and Collision Avoidance System (TCAS II) Installation.</td>
<td>The Certification Basis for the original product remains applicable to the certificated aircraft, CS25 elected to comply. Please refer to the EASA STC ST02761CH.</td>
<td></td>
<td></td>
<td>09/10/2009</td>
<td>Active</td>
<td>ST027558</td>
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<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>A300F4-606ST</td>
<td>Special Transports / Transports Spéciaux. (This STC, ref. ST25M002006-01) (Applicant reference), has been developed, reviewed, tested and found compliant with the Certification Basis to deal with non-Airbus/AS/350B-102. On case by case basis, this STC has to be beamanded to cover any new non...</td>
<td>As per EASA TCDS A.014 + CRI D-10 Issue 01. The Certification Basis for the original product and the following: Additional or alternative airworthiness requirements are applicable to the certificated aircraft.</td>
<td></td>
<td></td>
<td>12/10/2009</td>
<td>Active</td>
<td>ST027571</td>
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<td>PREMIER AIRCRAFT, LLC</td>
<td>FAA STC ST01951LA</td>
<td>Validation of FAA STC ST01951LA Installation of TFE731-4C engine part number: 3075900-1, per Honeywell Service Bulletin TFE731-72-7018 dated January 19, 2007 on the NSF50 basic aircraft configuration in accordance with Flight Test Associates Inc. Master Drawing List (MDL) No. PA-00100, Revision 0...</td>
<td>The Certification Basis for the original product remains applicable to the certificated aircraft. This certification approval involves a change to the certificated noise and/or emissions levels</td>
<td></td>
<td>Not Applicable</td>
<td>13/10/2009</td>
<td>Active</td>
<td>ST027600</td>
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<td>STC Holder</td>
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<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>MAX-VIZ Inc.</td>
<td>FAA STC ST01523LA</td>
<td>F900EX, MF50, MF90D</td>
<td>EASA Validation of FAA STC ST01523LA Installation of Max-Viz Enhanced Vision System (EVS) 1000 in accordance with Master Drawing List Document number 3038-00025 Revision 5, dated 14th October 2008 or later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revision...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List Document number 3038-00025 Revision 5, dated 14th October 2008 - Aeroplane Flight Manual Supplement Document No. 3038-00026, Revision A, Adated 16 December 2008 for the basic Mylere Faktion 50 aircraft configuration - Aeroplane Flight Manual Supplement Document No. 3038-00027, Revision A.</td>
<td>This design change is also applicable on the Mystere Faktion 50 designation as defined on EASA TCDS.A.062 but limited to the serial number aircraft MSN 286. For this MSN 286 aircraft the applicable EASA Flight Manual Supplement is the initial release of Document No.3038-00447 or later EASA approved revision...</td>
<td>13/10/2009</td>
<td>Active</td>
<td>10027612</td>
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<tr>
<td>AVIATION SUPPORT GmbH</td>
<td>REV. 1</td>
<td>AS 350 B3, AS 350 D</td>
<td>Installation of a 3D laser measurement system in an external cargo pod.</td>
<td>FAR 27 The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- ICA/ASE-06063 Revision - Instructions for Continued Airworthiness: MDL-ASE-06063 Revision Master Document List (Listen der gültigen Dokumente): FMS-ASE-90603 Revision - or later EASA approved revision - Flight Manual Supplement or later revisions of the above listed document(s) approved/accepted under the EASA system</td>
<td>None</td>
<td>13/10/2009</td>
<td>Active</td>
<td>10027603</td>
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<tr>
<td>AEROSPACE DESIGN AND ENGINEERING INGOLSTADT GmbH</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>A330-223</td>
<td>Introduction of a change from a 480/182Y dual class seating configuration to a 242/242Y passenger seating configuration.</td>
<td>Certification Basis in accordance with the EASA Type Certification dataSheet TCDS A.004 issue 16 with CS25 Amendment 4 for the areas affected by the change (plus revision to aircraft's TC basis for $25.785 (3) (c)(2) head impact criteria).</td>
<td>- Modification Summary, doc. No MD-1475 Revision D dated 15/10/09 - Aircraft Manual Supplement Document No: AMM-336-204-209 dated 01/07/09 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>03/12/2009</td>
<td>Active</td>
<td>10027603</td>
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<tr>
<td>GAMA ENGINEERING LTD.</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>F28 MARK 0070</td>
<td>Interior Refurbishment and System Upgrade</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and or emission levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- Change Level 1383 Release A, 10/08/2009 (per original certificate) - Continued Airworthiness TN1838-091 Release B, 10/12/2009 (per original certificate) - Certification Plan TN1838-028 Release A15 December 2009 - Flight Manual Supplement FMS1838-095 Issue 2 - EASA Terms of Approval 212.174 Issue 7, 1...</td>
<td>None.</td>
<td>26/11/2012</td>
<td>Active</td>
<td>10027637</td>
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<tr>
<td>GAMA ENGINEERING LTD.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>PA-E23-25 PA-23-230, PA-23-260</td>
<td>Installation of GamiN G500/G600 EFIS PFD/NHD</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. Non-significant The requirements for environmental protection and the associated certification noise and or emission levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List MLD-1067-02 Issue 1 - Change Statement MLD-1067-02 Issue 1 - Instructions for Continued Airworthiness ICA/1067-02 Issue 1 - Flight Manual Supplement AFM1067-02 Issue 1 for later revisions of the above listed documents approved by EASA</td>
<td>None.</td>
<td>21/10/2010</td>
<td>Active</td>
<td>10027677</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10027658</td>
<td>BEECHCRAFT CORPORATION</td>
<td>REV. 1</td>
<td>FAA TMD SA0253LA</td>
<td>B300/E300C</td>
<td>Installation of Belly Radome</td>
<td>FAR 23: The Certification Basis for the original product remains applicable to this STC approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- Installation in accordance with Master Data List 3003568671 Revision 0, dated September 14, 2009, or later approved revisions. - Operation in accordance with Airplane Flight Manual Supplement, 130-56031-327, September 2009, for 16,500 lb MTOW Configuration.</td>
<td>02/10/2011</td>
<td>Active</td>
<td>10027658</td>
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<tr>
<td>10027677</td>
<td>HELIKOPTER AIR TRANSPORT GMBH</td>
<td>(EASA,R.S.01499)</td>
<td>EC 135 T2 (CPDS) EC 135 T2+ EC 135 T1 (CDS)</td>
<td>Engineering Order EO 85-002, dated 21 September 2009 “Replacement of Helo Hoeka with Onboard Talon Systems”.</td>
<td>JAR 27 first issue - 04, September, JAR 27 Appendix C for Category A Operation, and JAR 27 Appendix B for operation under IFR; CS 21 for non-connection with CS 27.20(2)(ii) to comply acc. to CRI A-6. The Certification Basis for the original product remains applicable to this STC approval.</td>
<td>Installation Manual IM 85-002, dated 08.09.09; Onboard CMM 528-023-01, Rev. 15, dated 23.04.09; Indareni Siren CMM SL67-3, dated 22.08.05; AFM Supplement 85-002 T1 (CDS), dated 08.07.09; AFM Supplement 85-002 T2 (CPDS), dated 08.07.09; AFM Supplement 85-002 T2+, dated 08.07.09. Parts Overview 85-002-01.</td>
<td>Installation does not constitute operational approval. Approval of PCDS (Personal Carrying Device System) for external hoist operation must be granted by the competent authority.</td>
<td>20/10/2009</td>
<td>Active</td>
<td>10027677</td>
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<tr>
<td>10027678</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING GMBH</td>
<td>(EASA,R.S.03803)</td>
<td>A330-223</td>
<td>Airbus A330-200 B/C ISPSS Installation</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to STC approval: - CRI H-01: Special Condition H-01: Enhanced Airworthiness Programme - essayplane Systems - ICA on EWS. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- Engineering Bulletin, ref. doc. No. EB_M0650111 MB0_020 62009_A4BD_0429_04, Revision 1, dated 25th October 2009. Engineering Order, ref. doc. No. EO_M0650111 MB0_040 62009, Revision 2, dated 28th October 2009. Certification Program, ref. doc. No. CEP_M0650111_MGR_1 5642010 Revision D01, dated 20th October 2009.</td>
<td>This STC is limited to the following Serial Numbers: MSN 043, 432, 444, 456, 476, 493.</td>
<td>25/06/2010</td>
<td>Active</td>
<td>10027678</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10027707</td>
<td>AERO VODOCHODY AEROSPACE A.S.</td>
<td>REV 2</td>
<td>EMB-135BJ</td>
<td>EMB-135BJ</td>
<td>EFB class rev/Aero Bag Electronic Bag System Implementation</td>
<td>- Master Data List AM001-00-00-PA002 Rev. A &amp; B, 17 Sep 09-AMM Supplement AM001-00-00-LM001 Rev. A &amp; B, 30 Sep 09-AMM Supplement AM001-00-00-LM002 Rev. A &amp; B, 1 Oct 09, later revisions of the above listed documents approved by EASA</td>
<td>- Application of Electronic Flight Bag (EFB) Class 2 NAV, No external devices are approved to be connected to the display during movement/flight of the aircraft.</td>
<td>17/02/2013</td>
<td>Active</td>
<td>10027707</td>
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<tr>
<td>10027723</td>
<td>CESSNA ZURICH CITATION SERVICE GMBH</td>
<td></td>
<td>52S / 52SA / 52SE</td>
<td>52S</td>
<td>MINIGAR installation. By installing an Avionics miniGAR MK II and the MKIII according to the Installation Instructions OPM-009.</td>
<td>- Master Document List, Doc. No. MOL-009-ICA, Doc. No. ICA-008 or later revisions of the above listed documents approved by EASA.</td>
<td>S/N 525-0600 &amp; ON, 525A-0001 &amp; ON, 525B-0135 &amp; ON.</td>
<td>26/10/2009</td>
<td>Active</td>
<td>10027723</td>
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<tr>
<td>10027731</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td></td>
<td>FAA STC ST03705CH-0</td>
<td>GV-SP</td>
<td>Aircraft Serial No 5240 completion and cabin interior installation.</td>
<td>- Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52-2049146, Revision B, dated October 6, 2008, or later approved revisions. - Operation to be in accordance with Gulfstream Flight Manual Supplement GC52-2048044, Original Release, dated October 6, 2008, or later.</td>
<td>GV-SP Serial No. 5240 Only.</td>
<td>26/10/2009</td>
<td>Active</td>
<td>10027731</td>
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<td>STC Holder</td>
<td>Revision</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>BOMBARDIER INC.</td>
<td>REV. 1</td>
<td>TCCA STC SA05-60</td>
<td>BD-700-1A11</td>
<td>Installation: Complete Custom Aircraft Interior - ASN 9343</td>
<td>EASA TCDS No IM-A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. SA05-60 Issue 1, dated 14 June 2009-Aero Consulting Services Ltd Configuration Definition List No. E025009Rev. DB dated 31 May 2010-Bombardier's Airplane Flight Manual Supplement G-FM25004501 Revision NI- C-G029349000-PCFA Instructions for Continued Airworthiness or later revisions of t...</td>
<td>Restricted to ASN 9343.</td>
<td>04/06/2010</td>
<td>Active</td>
<td>10027762</td>
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<tr>
<td>EUROCOPTER UK Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL</td>
<td>EC135 T2+, EC135 P2+</td>
<td>Title: Installation of an External Camera and related ENG equipmentDescription: Installation of an External Camera and related ENG equipmentInstallation of an External Camera and related ENG equipment on EC UK change MHL/135/1198 as defined in Dnc. No.A3/1351198-001-11 Issue A and installation of the related ENG equipmentReference: MHL/135/1198 a...</td>
<td>JAR27 First Issue as per TCDS EASA R.009 The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement - EC135 FMS No. 53 Rev. 0 or later revisions approved by EASA/CA - Maintenance Requirement Document MR/135/1200-001 Issue A-Maintenance Requirement Document MR/135/1198-001 Issue C</td>
<td>None</td>
<td>30/10/2009</td>
<td>Active</td>
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<td>EASA STC No.</td>
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<td>Rev.</td>
<td>FAA STC</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>1002778</td>
<td>STIRLING DYNAMICS GMBH</td>
<td>REV 4</td>
<td>FAA STC</td>
<td>PC-12, PC-12/46, PC-12/47</td>
<td>Heavy Duty Floor for PC-12 Aircraft</td>
<td>Rev. 4: Transfer of STC</td>
<td>- Engineering Order BUPC12-E0-0001-01-02 Issue 01 Aircraft Flight Manual Supplement BUPC12-AM-5300-01-03</td>
<td>15/06/2022</td>
<td>Active</td>
<td>1002778</td>
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<tr>
<td>10027805</td>
<td>RAM AIRCRAFT, LIMITED PARTNERSHIP</td>
<td>REV 1</td>
<td>FAA STC</td>
<td>IO-520-B, -BB, -C, -CB</td>
<td>Certification Basis for the original product remains applicable to this certificate</td>
<td>- Certification Basis for the original product remains applicable to this certificate</td>
<td>Heavy Duty Floor P/N 017258 is limited to the PC-12 Aircraft in accordance with Bucher Flight Manual Supplement P/N 017270 and P/N 0172740-01; it is limited to from S/N 1010 onwards. The EASA STC is not applicable to the PC-12 Aircraft in accordance with Bucher Flight Manual Supplement P/N 017270-01</td>
<td>17/11/2009</td>
<td>Active</td>
<td>10027805</td>
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<tr>
<td>10027818</td>
<td>GAMA ENGINEERING Ltd.</td>
<td>REV 2</td>
<td>CAA UK RESPONSIBILITY</td>
<td>757-200</td>
<td>Installation of Garmin G580/G600 PFD/ND</td>
<td>- Master Data List MOD1100-02 Issue 4 - Change Statement C51100-02 Issue 2 - Instructions for Continued Airworthiness ICA1100-02 Issue 1 - Flight Manual Supplement AFMS1100-02 Issue 4.4</td>
<td>This change is limited to the configuration defined in Change Statement C511100-02 Issue 2 or later approved revision.</td>
<td>01/04/2010</td>
<td>Active</td>
<td>10027818</td>
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<tr>
<td>10027820</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV 1</td>
<td>757-200</td>
<td>205 Passenger LOPA</td>
<td>The Type and OSID Certification Bases (CB) for the original product are applicable to this certificate</td>
<td>- Major Data List MOD-1135-00001, Issue 1, 23/10/2003, -Modification Summary Sheet, MOD-1135-0001, Issue 1, 23/10/2003, -Key later revisions of the above listed documents, accepted to the EASA system.</td>
<td>The STC holder shall submit to the agency before 7th June 2020 final instructions of the instructions for continued airworthiness on the Electronic Wiring Interconnection System (EWIS) in accordance with the special condition (8-01) prior to installation of this change/repair, if it must be determined beforehand.</td>
<td>09/11/2021</td>
<td>Active</td>
<td>10027820</td>
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<tr>
<td>10027822</td>
<td>CRANFIELD AEROSPACE LTD</td>
<td>REV 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BAE JETSTREAM 3100</td>
<td>Installation of Seatom System</td>
<td>- Cranfield Modification No CaaM/jets3100/12870 Ype Design Definition Addendum C/D/DD/DD0054 Issue A</td>
<td>None</td>
<td>09/11/2009</td>
<td>Active</td>
<td>10027822</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>STC ref.</td>
<td>Models</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10027823</td>
<td>L-3 WESCAN INC.</td>
<td>REV.1</td>
<td>TC(4) STC: SH97-31</td>
<td>2. A335 E, F, F1, F2, N, NP, 1. A3350 B, B1, B2, B3, BA, D</td>
<td>MX-Series Cameras Installation (Structural Provisions Only)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAMS21446 revision 3 dated 4 October 2010 - Flight Manual Supplement requires the revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision)</td>
<td></td>
<td>04/02/2011</td>
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<td>10027824</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
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<td></td>
<td>CL-600-2B16</td>
<td>Installation of Honeywell Aircraft Flight Information System (AFIS)</td>
<td>EASA IM.A.023The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- RUAG Master Document List No. 076L-MML Revision N/C dated 03/10/2009 - RUAG AFMS No. 076L-AFMS Revision N/C dated 20/10/2009 - RUAG SoC ref. 076L-CN Revision N/C dated 23/10/2009.</td>
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<td>10027825</td>
<td>RUECKER AEROSPACE GmbH</td>
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<td>B777-200</td>
<td>Installation of 2 single B/E Aerospace DS5900 Oven Control Modules per B/E Aerospace Engineering OrderEO-353-0004-001 Rev. D. or later approved revision.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- B/E Aerospace Aircraft Maintenance Manual Supplement 43208-MA01- B/E Aerospace Component Maintenance Manual for basic part number 4329102</td>
<td></td>
<td>09/11/2009</td>
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<td>10027831</td>
<td>NORTHERN AVIONICS S.R.L.</td>
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<td></td>
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<td>A1095</td>
<td>NAV-396 &quot;Installation of Helicopter Egress Light System (HEEL)&quot;</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- RFM Supplement NAV-396 Rev. 0 or later revisions of the above document approved by EASA.</td>
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<td>09/11/2009</td>
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<td>1002786</td>
<td>BOMBARDIER AEROSPACE</td>
<td></td>
<td></td>
<td></td>
<td>CL-600-2B16</td>
<td>Optional Fuel Quantity Display in Kilograms</td>
<td>EASA IM.A.023The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- TC(4) STC No. SA09-53 Issue 1 dated 20 June 2009 - Aero Consulting Services Ltd. Modification Data Summary G336509Revision N/C or later approved revision.</td>
<td></td>
<td>11/11/2009</td>
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<td>1002787</td>
<td>KUERZI AVIATION AG</td>
<td></td>
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<td>LEARJET 55</td>
<td>Installation of Satellite Tracking System SAT-111</td>
<td>As specified in PCP-200931957, rev. A (or later approved revisions) The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
<td>11/11/2009</td>
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This STC is for structural provisions only. Functional and electrical aspects will require separate airworthiness approval. VNE limited to 126kts. VFR only. Category A operations prohibited.
<table>
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<tr>
<th>STC Holder</th>
<th>STC No</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>DOA AESGLOBAL LIMITED</td>
<td>10027883</td>
<td>REV. 2</td>
<td>A320-214</td>
<td></td>
<td>Introduction of Rockwell Colbas ePACES IFE System/Revision of the EASA STC 10027883 to remove EWIS limitation/Rev 2: Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to all certificated airframes.</td>
<td>- Modification Summary ref: Modification No. AES-320-1522, Issue 4 dated 21st March 2011 - Master Data List ref. doc. No. AES-320-1020/MDL; Issue 4, dated 23rd April 2010. Project Specific Certification Plan (PSCP), ref. doc. No. AES-TR-0690; Issue 4, dated 12th October 2009 - Certification Compliance R.</td>
<td>Not applicable</td>
<td>21/04/2021</td>
<td>Active</td>
<td>10027892</td>
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<td>BUCHAR LEICHTBAU AG</td>
<td>10027884</td>
<td>REV. 4</td>
<td>SEE AIRCRAFT ALLOCATION TABLE E043-04-12</td>
<td></td>
<td>16G NGS XL Stretchr PN 1120000- SeriesMajor Change to EASA STC 10027884 Revision 11 through - Addition of A315-100N, A320-200N, A321-200N, A321-200NX aircraft to the EASA STC 10027884 Aircraft Allocation Table E043-04-12 - Addition of 737-8 &amp; 737-9, 737-800 aircraft to the EASA STC 10027884 Aircraft Allocation Table E043-04-12</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: for details see below20, 1, 25, 27, 25, 25, 25, 27, 25, 30, 30, 25, 30, 30, 25, 30, 30, 25</td>
<td>- Master Data List E034-03-08 Rev. V dated 03.11.2023 - Aircraft Allocation Table E043-04-12 Rev. T dated 30.10.2023 - User Manual E017-07-09 Rev. 09 dated Nov 01/2023</td>
<td>- Par User Manual E017-07-09: This STC constitutes an approval for installation only for aeronautical serial numbers as stated in the Aircraft Allocation Table. For additional aircraft serial numbers, the requirements for full compliance have to be demonstrated through a design approval. No electrical / media...</td>
<td>10/11/2023</td>
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<td>10027894</td>
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<tr>
<td>SAFRAN ELECTRONICS &amp; DEFENSE SS UNIT - HELICOPTERS DATAMANAGEMENT INT SOLUTIONS &amp; SERVICES</td>
<td>10027889</td>
<td>REV. 1</td>
<td>EC 120 B</td>
<td></td>
<td>Installation of flight data monitoring and transmitting systems (a) - Installation of SafetyPlane/ Helicom V1 (PN4400) Rev 1: introduced name change from SafetyPlane to Helicom V1. (b) Addition of installation of Helicom V2 NUM (PN4700-E or PN4710-E (Satellite option)) Rev 2: introduced installation of Helicom V2...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): CRF F-01 Lithium Battery Installation (from Rev 2 onwards). The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended by the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>For SafetyPlane/Helicom V1/ (PN4400)- Design Document N EC120 - Design Document E-A-07,- Installation Manual N EC120 Installation Document E-A-07-: XDA N EC120 Supplement to Maintenance Manual E-A-05:- MDS; RF MS; --; For Helicom V2 (PN4700-E or PN4710-E (Satellite option)) - Design Document: ALDEC</td>
<td>Prior to installation of this change/repair it must be determined that there is an interrelationship between this installation and any other previously installed change and/or repair and no adverse effect upon the airworthiness of the product.</td>
<td>27/02/2020</td>
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<td>UNAIR</td>
<td>10027898</td>
<td>REV. 2</td>
<td>MF900</td>
<td></td>
<td>Installation of a Flight Data Recorder</td>
<td>The Certification Basis for the original product remains applicable to all certificated airframes.</td>
<td>- Master Data List Document MDL-019-07 Revision 5, dated 09th October 2005 - Aerospatiale Flight Manual Supplement UA0-AMF5-08-06, Revision 2 dated 07th August 2006</td>
<td>This design change is applicable on the basic Mystere Falcon 900 series aircraft, as per MF900 designation on EASA TCDS A.062</td>
<td>13/11/2023</td>
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<td>STC Holder</td>
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<td>Date of Issue</td>
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<td>10027900</td>
<td>FAA STC 01712A</td>
<td>747-400, 747-400F</td>
<td>Installation of Teledyne Controls Wireless Ground Link - Quick Access Recorder (WQAR) Rev. 2 adds 747-8-8F models.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List M DL224360-747, Revision H, dated 01 October 2021 &lt;(v.&lt;.v.) or later</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/04/2022</td>
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<td>10027902</td>
<td>AERONAUTICAL</td>
<td>BELL 407</td>
<td>Expanded Area Instrument Panel</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>AA-97129 revision F dated 28 May 2009 - Drawing List, Expanded Instrument Panel for Bell Helicopter Textron 407 Model Helicopters; AAT1730 revision F dated 18 April 2008 - Installation Instructions for Expanded Instrument Panel AA-02052 revision A dated 30 July 2004 - Instructions for Continuation/Agreement.</td>
<td>None.</td>
<td>01/12/2009</td>
<td>Active</td>
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<td>10027904</td>
<td>FLYHT AEROSPACE</td>
<td>HAWKER 900XP</td>
<td>AFIRS 220 AFIRS 228 Automated Flight Information Reporting System. Installation of Teledyne Global Satcom IP (AFIRS) 228 System installation is FAR 25 including Amendments 25-1 through 25-97 and FAR 25-88 at Amendment 25-102.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>AFIRS 220 System Installation - Installation of AFIRS 220 system must be in accordance with NAVWX/Weather Services Ltd (“FLYHT”), Installation Instructions 151-0032, Rev. E, Transport Canada (TCCA) approved 18 October 2010, (order) TCCA approved.</td>
<td>None.</td>
<td>29/08/2012</td>
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<td>10027913</td>
<td>AERONAUTICAL</td>
<td>BELL 407</td>
<td>Vertical Fin Cap with LED Light Kit</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>AA-04133 revision A dated 18 February 2009 - Drawing List, Vertical Fin Cap with LED Light Kit K3 AAT97130 revision F dated 18 April 2008 - Installation Instructions for Expanded Instrument Panel AA-05111 no revision dated 23 February 2009 - Instructions for Continuation/Agreement.</td>
<td>None.</td>
<td>01/12/2009</td>
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<td>10027954</td>
<td>AVIATION TRADERS LIMITED/EUROPE S.A.S.</td>
<td>BAE 146 SERIES 200</td>
<td>BAE 146 VIP Cabin Interior + Revision due to the removal off the EWIS limitation on EASA STC 10027954Rev 3. Transfer of STC</td>
<td>Master Document List MDL-11195-0001 Issue 4/1/ Approval of the above listed document (s) approved accepteunder the EASA system - Modification Summary Sheet MOD-11195-0001 Issue 4 - Master Document List MDL-11195-0001 Issue 4 - Compliance record CR-11193-0001 Issue 2 - Enhanced Zone.</td>
<td>09/11/2021</td>
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<td>10027985</td>
<td>DANSK FLY ELEKTRONIK ApS</td>
<td>F172P, F172K, F172L, F172M, F172N, F172D, F172E, F172F, F172M.</td>
<td>ASPEN EFD1000 installation</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/11/2009</td>
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<td>10027989</td>
<td>SOCIETE CETUXE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td>DHC-8-100, -200, -300</td>
<td>Installation of ELT KANNAD 406 AF-COMPACK or ARTEX ME406</td>
<td>Only applicable for aircraft and/or products approved in accordance with the conditions as defined in Design Description, Integration of AD-AFIS-260, Doc. No. DES-AFC-887, Rev. 2 dated 17-05-2009, or later approved revision</td>
<td>23/11/2009</td>
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<td>10028007</td>
<td>DEVORE AVIATION CORPORATION OF AMERICA</td>
<td>FAA STC SR00507RC</td>
<td>Installation of Talir/Rotor, Forward Facing Recognition and Fuselage/leading Edge lights in accordance with DeVore Aviation Corporation Drawing Number 44005001. This installation is based off FAA STC SR00507RC</td>
<td>None</td>
<td>24/11/2029</td>
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<td>LAST REFUE LIT</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>FRJ72, R172K, 182, 208</td>
<td>Approval of Flight Manual Supplements for last Refuge Ltd Wing strut mounted camera installation.</td>
<td>BCAR Section K</td>
<td>Flight Manual Supplements.--LAST REFUE LIT. WING STRUT MOUNTED CAMERA SYSTEM (MARK II) FOR CESSNA AIRCRAFT SUPPLEMENT in CESSNA FRJ72 Reims Rocket, WITH CONTINENTAL IO-360 210HP, FLIGHT MANUAL, AEE SUPPLEMENT 1, ISSUE 2, dated 20 August 2009. LAST REFUE LIT. WING STRUT MOUNTED CAMERA SYSTEM (MARK II).</td>
<td>None</td>
<td>26/11/2009</td>
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<td>10228074</td>
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<td>ALDAUNIA S.R.L.</td>
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<td>BM No. 020/AL/2008</td>
<td>Installation of a &quot;Ginevri&quot; incubator Type &quot;Baby Shuttle Intensive Care&quot;</td>
<td>The Certification Basis for the original product remains applicable to this STC certificate/approval.</td>
<td>RFM Supplement No. FMS/ AL/2020 Rev. 0</td>
<td>None</td>
<td>26/11/2009</td>
<td>Active</td>
<td>10228020</td>
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<td>GAMA ENGINEERING LTD</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>PA-34-220T, PA-34-200, PA-34-200T.</td>
<td>Installation of Aspen EFD1000 PFD as add on</td>
<td>The Certification Basis for the original product and the following additional or alternative avionics requirements are applicable to this certificate/approval</td>
<td>EASA approved Master Document Ltd, Doc. No. MDL1100-01, Issue 1, dated 04-Nov-2009, or later approved</td>
<td>None</td>
<td>Only applicable for airframe/ preconditions as defined</td>
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<td>26/11/2009</td>
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<td>10228020</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>650 (CITATION VI)</td>
<td>Software upgrade of installed Honeywell GNS-XL or GNS-XLS FMS system for RNAV operation</td>
<td>CS 25 amendment 4 The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Engineering Order 2050529-EO Issue 2 from 27 November 2006, list of STC documents &quot;2050529 STC&quot; Issue 2 from 27 November 2006, AFM Supplement 2050529-AM-A Issue 2 from 27 November 2009 for GNS-XL, FMS Supplement 2050529-AM-B Issue 2 from 27 November 2009 for GNS-XLS Declaration of Compliance from 27...</td>
<td>None</td>
<td>Limited to: SN 650-7001 thru 650-7199 The STC applies only for Single or Dual FMS GNS-XL or GNS-XLS configuration, approved for B-RNAV operation. Interfering with other FMS is not approved within this STC.</td>
<td>30/11/2009</td>
<td>Active</td>
<td>10228009</td>
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<tr>
<td>GAMA ENGINEERING LTD</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>182 SERIES</td>
<td>Installation of Aspen EFD1000 PFD</td>
<td>CS-23 The Certification Basis for the original product and the following additional or alternative avionics requirements are applicable to this certificate/approval.</td>
<td>Master Data List MDL1119-01 Change Statement CS1119-01 Instructions for Continued Airworthiness, IC.A1119-01 Flight Manual Supplement AFM1119-01 Issue 1 for later revisions of the aforesaid documents approved by EASA</td>
<td>None</td>
<td>None</td>
<td>30/11/2009</td>
<td>Active</td>
<td>10228009</td>
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<td>SABENA TECHNICS BOD</td>
<td>A340-311/312/313</td>
<td>IAGOS Modification This STC installs the integration of routine aircraft measurements into Global Observing System. Thesidual Release of the STC was approving the technical provisions installing the IAGOS system. This Revision 1 approved the activation of the IAGOS system, and updating some of the li...</td>
<td>The Modification Certificate (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Modification Approval Sheet, ref.: S-00-2071, Issue 9 dated 20 February 2014 Aircraft Flight Manual Supplement, ref.: AFM-5-F4030-01 Rev 04, dated 07 October 2013 Instruction for Continued Airworthiness, ref.: ICAI-1119-01 Flight Manual Supplement AFM1119-01 Issu...</td>
<td>None</td>
<td>The modification is applicable to the following aircraft Serial Numbers: MSN 221, 304, 327, 335, 347, 352, 354, 355, 377, 379, 390, 433, 434, and 447 for an installation of this design change it must be determined if the interrelationship between this design change and any other previously installed...</td>
<td>27/02/2014</td>
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<td>STC Holder</td>
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<td>AVIONICS INTEGRATION &amp; ENGINEERING INCORPORATION AG  (AIEC)</td>
<td>Installation of Artex ELT C406-02 (Coiband).</td>
<td>The Certification Basis for the original product and the following additional/alternative airworthiness requirements are applicable, if certified/approval as specified in Certification Program 900-080, rev. A (or later approval revisions). The certified noise and/or emissions levels of the...</td>
<td>None</td>
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<td>01/12/2009</td>
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<td>AVIONICS INTEGRATION &amp; ENGINEERING INCORPORATION AG  (AIEC)</td>
<td>Installation of ART 2000 Weather Radar with MAD 850 MD2 (Honeywell).</td>
<td>The Certification Basis for the original product and the following additional/alternative airworthiness requirements are applicable, if certified/approval as specified in Certification Program 912-080, rev. A (or later approval revisions). The certified noise and/or emissions levels of the...</td>
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<td>AVIONICS INTEGRATION &amp; ENGINEERING INCORPORATION AG  (AIEC)</td>
<td>Installation of SSFDR F1000 (L-3 Communications).</td>
<td>The Certification Basis for the original product and the following additional/alternative airworthiness requirements are applicable, if certified/approval as specified in Certification Program 948-080, rev. A (or later approval revisions). The certified noise and/or emissions levels of the...</td>
<td>None</td>
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<td>AVIONICS INTEGRATION &amp; ENGINEERING INCORPORATION AG  (AIEC)</td>
<td>Installation of EGPWS Mk VIII (Honeywell).</td>
<td>The Certification Basis for the original product and the following additional/alternative airworthiness requirements are applicable, if certified/approval as specified in Certification Program 911-080, rev. A (or later approval revisions). The certified noise and/or emissions levels of the...</td>
<td>None</td>
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<td>AVIONICS INTEGRATION &amp; ENGINEERING INCORPORATION AG  (AIEC)</td>
<td>Installation of CVR-308 (JASIC).</td>
<td>The Certification Basis for the original product and the following additional/alternative airworthiness requirements are applicable, if certified/approval as specified in Certification Program 851-049, rev. A (or later approval revisions). The certified noise and/or emissions levels of the...</td>
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<td>PATS AIRCRAFT L.L.C.</td>
<td>Alteration of Fire Protection Systems, Lower Cargo Compartments. This STC revises and supersedes LBA STC LBA-21NE1.TA0682. The revision is issued to reflect the transfer from Hollingshead International to Decrease Aerospace PATS Aircraft. Rev 1. Added reference to MMEL Supplement.</td>
<td>The Certification Basis for the original product as amended by the following additional/alternative airworthiness requirements - CRJ A01, 9737-730, Issue 9, 13 February 1998. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are...</td>
<td>Hollingshead Master Data List 17106 Rev. M dated August 01, 2007 FAA Airplane Flight Manual Supplement 20629 Rev. NC/Instructions for Continued Airworthiness 20689 Rev. 09/2010; MMEL Rev. G dated 03 March 2010; Other revisions of the above listed document(s) approved.</td>
<td>Prior to installation of this change/repair it must be determined that this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/04/2021</td>
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<td>PATS AIRCRAFT L.L.C.</td>
<td>REV 2</td>
<td>ST01184LA</td>
<td>737-800</td>
<td>Alteration of Fire Protection Systems, Lower Cargo Compartments Rev. 0. This STC revises and supersedes LBA STC LBA-21NE1.TA0511. This revision was issued to reflect the transfer from Hildinghed to International to Decare Aerospace PATS Aircraft. Rev. 1 adds an additional metered fire bottle to both the</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated criteria for noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Configuration 1 per MDL 2544-01, Rev G dated 14 October 2006 - Configuration 3 Non-ETOPS Converted Freighters per MDL 2544-01, Rev K dated 22 July 2020 and Configuration Matrix 8148-26-CFGX-001 Revision 18 August 2020-</td>
<td>13/04/2021</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>(EASA.A.S.03901)</td>
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<td>Installation of a 2-blade MTV-15-B/203-58 propeller on Maule M-4 Models MX-7-180, MX-7-180 A, MX-7-180 B, MX-7-180 C, MX-7-180D, MX-7-180C, MX-7-180, and MX-7-180A</td>
<td>FAR-227. The Certification Basis for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the certificated noise and/or emissions levels</td>
<td>- Airplane Flight Manual Supplement E-1780, Original Issue dated 09. September 2009 or later approved revision - Instruction for Continued Airworthiness E-1782, Original Issue dated 09. September 2009 or later approved revision - Installation Instruction E-1781, Original Issue dated 09. September 2009 or later approved revision</td>
<td>None</td>
<td>01/12/2009</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>(EASA.A.S.03369)</td>
<td>FAA STC SA5357SW-D</td>
<td>PA-28-201T PA-28-181, PA-28-236, PA-28-161, PA-28-201</td>
<td>Installation of S-TEC Manual Electric Trim System, Model ST-156 (14 VoltSystem)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product unchange and remain applicable to this certificate/approval.</td>
<td>Applicable revisions: FAA Approved S-TEC Bulletin No. 258, dated 21-Nov-84. Initial Issue, or later revision FAA Approved S-TEC Master Drawing List No. 92508, dated 21-Nov-84. Initial Issue, or later revision FAA Approved Supplement to Pilot’s Operating Handbook and/or FAA Approved Airplane Flight Manual</td>
<td>Compatibility of this modification with other previously approved modifications must be determined by the installer. The Approval of the installation of this system is not applicable to all specific aircraft having other previously approved modifications, unless is determined by the installer that the...</td>
<td>01/12/2009</td>
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<tr>
<td>JET AVIATION AG</td>
<td>GULFSTREAM GIV</td>
<td>Mio Cabin Door</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>(EASA.A.S.03435)</td>
<td>A330-243</td>
<td>A330-200 - VIP Cabin Interior Completion</td>
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- Configuration 1 per MDL 2544-01, Rev G dated 14 October 2006 - Configuration 3 Non-ETOPS Converted Freighters per MDL 2544-01, Rev K dated 22 July 2020 and Configuration Matrix 8148-26-CFGX-001 Revision 18 August 2020- | 13/04/2021 | Active | 10226502 |
- Airplane Flight Manual Supplement E-1780, Original Issue dated 09. September 2009 or later approved revision - Instruction for Continued Airworthiness E-1782, Original Issue dated 09. September 2009 or later approved revision - Installation Instruction E-1781, Original Issue dated 09. September 2009 or later approved revision | Compatibility of this modification with other previously approved modifications must be determined by the installer. The Approval of the installation of this system is not applicable to all specific aircraft having other previously approved modifications, unless is determined by the installer that the... | 01/12/2009 | Active | 10226507 |
10028136  
SAVANNAH AIR CENTER, L.L.C.  
F AA STC No ST03818AT  
BOMBARDIER BD-700-1A10  
Complete Custom Aircraft Interior - ASN 9287.  
EASA MA.309 and EASA MOC CRI F.011The Certification Basis for the original product remains applicable tois certificate/approval. The certified noise and/ or emissions levelsof the original product are unchanged and remain applicable to this certificate/approval.  
EASA STC No ST03818AT dated 29 October 2008 FAA SuC ref. ST03818AT dated 30 November 2009 Savannah Air Center Master Drawing List. XRI0287-MDL-2500001. Rev Adated 14 October 2009 or later approved

10028137  
MOTORFLUG BADEN-BADEN GmbH  
REV. 4  
P.180 AVANTI II  
The Certification Basis for the original product as amendedby the following additional oralternative airworthiness requirements:CS-23, 1st issue/CS23.301, 23.303, 23.305, 23.307, 23.321, 23.341, 23.351, 23.562 (seatonly), 23.575, 23.601, 23.603, 23.605, 23.607, 23.608, 23.611, 23.613, 23.629, 23.853...

10028151  
RUAG AEROSPACE SERVICES GmbH  
REV. 1  
EMB-135BJ  
The Certification Basis for the original product and thefollowing additional or alternative airworthiness requirements are applicable to its certificate/approval:CS-27A, 1st issue/CS27A.052 plus -CRH H 11324 “EWS”-CRH-F.28-13304 EFB certificated noise and/or emissions levels of theoriginal product.

10028164  
AKV, Inc.  
REV. 1  
FAA STC SR00631DE-D  
Dual Engine Cycle Counter Installation of AKV, Inc. Dual Engine Cycle Counter BK117XCC systemised to acquire engine dataacross all existing engine speedinputs for gasproducer and free turbine/Inliner. The recommendation for approval is given based on a reworked performed inaccordance with the...

10028167  
STANDAR D AERO  
FAA STC ST01991IC  
MF50  
EASA Validation of FAA STC ST01991IC installation of an L3 Communications (Fairchild) Model F1000 Flight Data Recorder (FDR) System inaccordance with Garrett Aviation Services/Drawing List. Doc. Number 1002382 Revision H, dated 04th June 2004 or later revisions approved by EASA inaccordance with EASA2...

10028171  
APICAL INDUSTRIES, Inc/dba/DART AEROSPACE  
REV. 1  
FAA STC SR02244LA  
MBB-BK117 C-2, MBB-BK117 D-2  
Installation of Emergency Float Kits.  
The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental/and the associated certified noise and/or emissions levels ofthe original product are unchanged and remain applicable to this certificate/approval.

13/09/2018  
04/12/2009  
11/11/2013  
30/06/2010  
13/06/2018  
07/12/2009  
24/06/2018
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<th>Aircraft</th>
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<th>STC No</th>
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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tr>
<td>R44</td>
<td>AVONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>AERODISA S.L.</td>
<td>REV. 1</td>
<td>Falcon 2000 Pocket Door Installation for Falcon 2000m according to Aerodata Master Data List AED2007-09M MDL Revision 3: dated 04th December 2009 or later revisions approved by EASA</td>
<td>EASA CS-25 amendment 1 plus EASA Certification Review Item (CRI) D-01Issue 2 &quot;Installation of Interior Door&quot; for those areas, systems, parts and appliances of the original Falcon 2000 design affected by this design change.</td>
<td>-Pocket Door Installation for Falcon 2000m according to Aerodata Master Data List AED2007-09M MDL Revision 3: dated 04th December 2009 or later revisions of this list are approved by EASA- EASA Certification Review Item (CRI) D-01Issue 2 &quot;Installation of Interior Door&quot; for those areas, systems, parts and appliances of the original Falcon 2000 design affected by this design change.</td>
<td>21/12/2009</td>
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<td>B. B.90, C.90, H.90</td>
<td>ISEI</td>
<td>R44, R44 II</td>
<td>Title: Installation of Safetyplane Equipment Description: Installation of a system for tracking a set of flight and system parameters during operation of the rotorcraft. The modifications based on ESB change n. R44 as defined in the Master Document LN01-R44-Reference Documents E-A-02, dated 19.11.2.</td>
<td>CS-23 Addt. 17The Certification Basis for the original product remains applicable to this certificate/approval The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval</td>
<td>-Installation Instructions, R64-R44-Installation Document E-A-02- Instructions for Continued Airworthiness, R44-Maintenance manualsupplement E-A-02- N2-R44-Approval Change Document E-A-02- N3-R44-Experience Feedback E-A-02- R44-R44-Safety Assessment document E-A-02- NS-R44-Design Document E-A-22v...</td>
<td>None</td>
<td>08/12/2009</td>
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<td>A. C.90A, C.90GT, C.90GTI</td>
<td>BOMBARDIER INC.</td>
<td>BD-700-1A11</td>
<td>Installation: Complete Custom Aircraft Interior - ASN 9231</td>
<td>EASA TCC No MA-009The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>TCCA STC No 5555801 Issue 1 dated 14 June 2008 TCCA 5400 Rev 09-1021 dated 2 December 2009 Aero Consulting Services Ltd. Configuration Definition List E25000 Revision CC-Aero Consulting Services Ltd. Aircraft Flight Manual Supplement FM25004501 Revision N/C or later approved revision DOC2001000-F...</td>
<td>Restricted to ASN 9231</td>
<td>07/12/2009</td>
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<td>10028220</td>
<td>NSE INDUSTRIES S.A.</td>
<td>REV. 2</td>
<td>2 - 206 1 - L206E, L206F, U206G 1 - L206A, L206B, U206C 1 - TU206E, TU206G, TU206F 1 - TU206B, TU206C, TU206D 1 - TP206A, TP206B, TP206C 1 - P206C, P206D, P206E, P206H, T206H 1 - 208H, P208A, P208B</td>
<td>STC 10028220 revision 2 for STC DTS and DTS-I/2 Cessna 206 - Installation/Tracking System Description: Change of Corporate Name and location, replacement of DT9112 documentation reference, addition of TCDS USA AME to reflect all modularized incorporated of MC 1004139 and removal of associated tools...</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval...</td>
<td>FAA STC ST5220D-00 Rev. 0</td>
<td>5D-00/DTS-I/2-006 Rev. 00 dated 20/06/2010/EASA Certification basis CT52 I/2-036 Rev. 00 later revisions of the above listed documents approved by EASA...</td>
<td>The approval holder shall fulfill the obligations of Part 21, Paragraph 2A.105. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effects upon the a...</td>
<td>01/08/2013</td>
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<td>10028237</td>
<td>VAN REMSDUK ROTTERDAM B.V.</td>
<td>REV. 2</td>
<td>B747-400F</td>
<td>NXE-0506 Special Cargo Container 125” X 196” with part number 136.020.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval...</td>
<td>NVV-H+N Control and Loading Manual supplement RRMS 15 Issue 2CMM/IPL Container 136.020 doc. Nr RRMS 10 issue 1 or later revisions of the above listed documents approved by EASA...</td>
<td>NXE-0506 Special Cargo Container 125” X 196” with part number 136.020...</td>
<td>The location of the cargo container is limited to the positions mentioned in the NVV-H+N Control and Loading Manual supplement RRMS 10...</td>
<td>10/12/2009</td>
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<td>10028238</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>A318-112</td>
<td>A318-112 - Elite 11 - VIP Cabin Completion</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval...</td>
<td>- Supplemental Type Definition Document, ref. Doc. No STD00-090/01M, issued 02, dated Apr. 23/2010; - Airplane Flight Manual Supplement, ref. Doc. No. A-00-09/053-AFMS 04 rev. 01, dated Nov. 25/2009; Instructions for Continued Airworthiness, ref. Doc. No. A.00-09/053-ICA, issue 04, dated Mar. 30/2011...</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval...</td>
<td>Not applicable...</td>
<td>26/04/2010</td>
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<td>10028246</td>
<td>SOCATA SAS</td>
<td>REV. 2</td>
<td>TBM 700 N</td>
<td>Installation of a Honeywell HF/KHF 1050 communication system on aircraft TBM 700 N equipped with Garmin 1000 Glass Cockpit System - Update for CA inclusion...</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval...</td>
<td>Daher-Socata A/S Instructions for Continued Airworthiness ref.200 DIMNSTC01PEEB 0, original issue dated July 2013-Daher-Socata A/S Modification File No STC 23.001.09, edition 1, dated 09/12/2009...</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval...</td>
<td>None...</td>
<td>26/06/2010</td>
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<td>10028267</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST01796V-0</td>
<td>GV-SP</td>
<td>Aircraft Serial No 5246 completion and cabin interior installation...</td>
<td>Installation in accordance with Gulfstream Aerospace Master Drawing List No. GC52 204M047, Original Release, dated December 7, 2009, or later...</td>
<td>Gulfstream Aerospace Master Drawing List No. GC52 204M047, Original Release, dated December 7, 2009, or later...</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval...</td>
<td>GV-SP Serial No 5246 Only...</td>
<td>14/12/2009</td>
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<td>10028268</td>
<td>DASSAULT FALCON JET - WILMINGTON CO RP</td>
<td>FAA STC ST01796V-0</td>
<td>P900EX</td>
<td>EASA Validation of FAA STC ST5220D-00 Rev. 0 for Dassault Falcon Jet Wilmington Master Drawing List, 197900584 issued 09/2006 or later revisions approved by EASA...</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval...</td>
<td>Dassault Aviation Certification (CA) 1979002-00 dated 18/04/2006 or later revisions of the above listed documents approved by EASA...</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval...</td>
<td>This design change is applicable to Dassault Aviation Mystere Falcon 900 aircraft which have previously been equipped with Dassault Aviation Modification M3000 (Falcon 900EX definition)...</td>
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<td>Date of Issue</td>
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<td>10028306</td>
<td>STIRLING DYNAMICS GMBH</td>
<td>ERJ170-200STD, ERJ170-200LR, ERJ170-100STD, ERJ170-100LR</td>
<td>Installation of a removable stretcher</td>
<td>Rev 4: Transfer of STC</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements</td>
<td>Prior to installation of this change/repair it must be determined that the interchangeability between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/06/2022</td>
<td>Active</td>
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<td>10028318</td>
<td>STG AEROSPACE Inc.AKA SAFTGLO, Inc.</td>
<td>FAA TCDs T00001SE</td>
<td>Installation of S4Tigo, Inc photoluminescent floor proximity emergency escape path marking system</td>
<td>None</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>None</td>
<td>17/12/2009</td>
<td>Active</td>
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<td>10028319</td>
<td>AIRBUS S.A.S.</td>
<td>A319-115</td>
<td>ACJ Centre Modification CJ0300 - &quot;VIP Cabin Interior - AJC</td>
<td>ACJ</td>
<td>Centre Modification CJ0392 - &quot;Private to Public Conversion&quot;, Correction of design defect identified: use of additional seating positions during TTOL</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval and remains applicable to this certificate/revision</td>
<td>FAA STC ST03844ATSTG report AC37-2 Master Drawing List rev. DAFM supplement AC37-11 dated May 5, 2009-12-16 Instruction for Continued Airworthiness AC37-20 dated March 16, 2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/G of 15/06/2022.</td>
<td>04/07/2014</td>
<td>Active</td>
<td>10028319</td>
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<td>10028330</td>
<td>TATENHUI AVIATION Ltd.</td>
<td>BEECH 200</td>
<td>Design data to provide European approval of FAA STC: S5254NM</td>
<td>Master Data List TA</td>
<td>T</td>
<td>TAD 014/09 Rev B Instruction for Continued Airworthiness ICA 01409 Rev B Flight Manual Supplement 014/09 Rev Aor later revisions of the above listed documents approved by EASA</td>
<td>Serial number BB-877 only</td>
<td>17/12/2009</td>
<td>Active</td>
<td>10028303</td>
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Note: For the full list and specific details, please visit the EASA website at easa.europa.eu.
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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>S-TEC CORPORATION</td>
<td>Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-853 (28 Volt System)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Applicable revisions FAA Approved S-TEC Bulletin No. 953, dated 08-Mar-99. Initial Issue, or later revision -FAA Approved S-TEC Master Drawing List No. 921151, dated 08-Mar-99. Initial Issue, or later revision -FAA/DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, Pt N 858...</td>
<td>Compatibility of this modification with other previously approved modifications must be determined by the installer. The Approval of the installation of this system is not applicable to specific aircraft having other previously approved modifications. unless it is determined by the installer that the...</td>
<td>17/12/2009</td>
<td>Active</td>
<td>10028331</td>
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<tr>
<td>FLYING COLOURS CORP.</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>EASA.IM.A.023 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval EASA CRC No. H-01 The requirements for environmental protection and the associated certificate/noise and/or emissions levels of...</td>
<td>TCCA LSTC No. C-LSA09-326/D Issue 1 dated 24 November 2009. Aero Consulting Services Ltd. Modification Data Summary G28800 Revision G. Flying Colours Corp. Aircraft Flight Manual Supplement for the CompleteCustom Aircraft Interior G288091 Revision NC or later approved revisions. Flying Colours Corp. Inc...</td>
<td>Restricted to ABN 8092</td>
<td>14/12/2011</td>
<td>Active</td>
<td>10028338</td>
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<tr>
<td>BOMBARDIER INC.</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9320</td>
<td>EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Transport Canada SoC reference C-09-0275 dated 08 December 2009. TCCA STC No. SA4-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D952000 Revision DQ dated 08 December 2009. Instructions for Continued Airworthiness GC93200000-IFCA or later revisions of the above listed documents approved by EA...</td>
<td>Restricted to ABN 9320 and...</td>
<td>18/12/2009</td>
<td>Active</td>
<td>10028344</td>
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<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>Special Transports / Transports Spéciaux This STC ref. ST03M0020 ed.01 (Applicant reference) has been developed, reviewed, tested and found compliant with the Certification Basis to deal with non-Airbus sub-assemblies transport. On case by case basis, this STC had to be amended to cover any new non-A...</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval</td>
<td>Change Approval Sheet, ref. ST03M0020 ed.01 or later revisions of the above...</td>
<td>None</td>
<td>18/12/2009</td>
<td>Active</td>
<td>10028350</td>
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<td>EASA STC No</td>
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<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10028355</td>
<td>STC Holder: NORTHPOINT AVIATION</td>
<td>A320-232</td>
<td>Installation of New Rear Observation Seat in Textron 206\Rev. 3: Transfer from &quot;Fifty-Ten-Yancey, LLC&quot; to &quot;DV Aviation, Inc.&quot;</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise/and or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Soyby LLC Master Drawing List 910-100, Rev. N, dated 23 August 2012 - Service Instruction No. 910-01, dated 19 March 2009.</td>
<td>Weight and Balance: Equipment Removal, Weight (lbs), Station(0), 2nd Row (Seat), 75, 78.6 (Seat, 3rd Row (Berth), 38, 105 Equipment Installed, Weight (lbs), Station (in), Seat Assembly with Armrests, 34, 81.7 Base Plate Assembly, 23.2, 81.9 Seat Rail Assembly, 25.3, 81.7 Seat Rail Attachment.</td>
<td>04/03/2023</td>
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<tr>
<td>10028359</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A320-232</td>
<td>Humidification and Zonal Dryer System Installation: (project description document PD-AA3201-2100)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List doc. No MDL-A3201-2100-1 Rev D dated 25/02/2010 - Instructions for Continued Airworthiness doc. No ICA-A3201-2100 Rev C dated 24/02/2010 - Declaration of Compliance Form AA-DO-77 - Rev B dated 03/03/2010 with reference to Master Data List MDL-A3201-2100-1 Rev D latter revisions of the...</td>
<td>According to doc. No ICA-A3201-2100 as listed under associated technical documentation.</td>
<td>04/03/2010</td>
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<tr>
<td>10028361</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>A320-232</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9535</td>
<td>EASA TCDS No IMA-009The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- TCDA STC No. SA05-60 Issue 1 dated 14 June 2005 - TCDA STC ref. C-09-026 dated 15 December 2009 - Aero Consulting Services Ltd. Configuration Definition List E025000Revision CF - Aero Consulting Services Ltd. Airplane Flight Manual Supplemental-FMD25004901 Revision NIC or later approved revision. - GC9353.</td>
<td>Restricted to ASN 9535</td>
<td>21/12/2009</td>
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<td>10028365</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A320-232</td>
<td>Interior Outfitting (see project description document PD-AA01-2500-1) The limitation for use of Supplemental Water Tanks has been removed at Rev 1 of EASA STC 10028365.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval EASA Supplemental Type Certification Basis CRI A-2301-001 Issue 2The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval...</td>
<td>- Master Data List doc. No MDL-A401-2500-1 Rev 1 dated 05/01/2010 - Instructions for Continued Airworthiness doc. No ICA-A401-2500 Rev C dated 17/12/2009 - Airplane Flight Manual Supplement doc. N AFMS-A401-2500-1 Rev A dated 04/12/2009 - Declaration of Compliance Form AA-DO-77 Rev A dated 08/11/2010 with reference to Master Data List MDL-A401-2500-1 Rev 1 latter revisions of the...</td>
<td>Restricted to MSN 3379</td>
<td>08/01/2010</td>
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<td>10028396</td>
<td>MAGNIFICENT MRO AS</td>
<td>A320-232</td>
<td>Installation of Galley G1 and Galley GS. Revision of the EASA STC to remove EWIS: Imitation Revision: Transfer of Certificate to Magnetic MRO AS</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI A-2301-001 Issue 2The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval...</td>
<td>- Drawing &amp; Document Control Report doc. No. 00DC008, dated 21/12/2009 - Drawing &amp; Document Control Report doc. No. 00DC009, dated 21/12/2009 - Declaration of Design and Performance doc. No. 00D0301, Issue B, dated 18/12/2009 - Declaration of Design and Performance doc. No. 00D0302, Issue B, dated 18/12/2009.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/09/2019</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
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<td>Associated Technical Documentation</td>
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<td>10028374</td>
<td>CANARD AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST01331CH</td>
<td>DHC-8-201, DHC-8-202</td>
<td>Installation of Honeywell Mark VII Enhanced Ground Proximity Warning System to replace the existing Mark II Ground Proximity Warning System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation must be in accordance with Canard Aerospace Document Control Drawing No. 40094, Revision O, dated December 12th, 2010. Operation must be performed in accordance with FAA approved Airplane Flight Manual Supplement, Canard Aerospace: 34-3250-24, Revision A, dated September 24th, 2012</td>
</tr>
<tr>
<td>10028376</td>
<td>AERO DYNAMIX, Inc.</td>
<td>FAA STC SR09480RC</td>
<td>MD600</td>
<td>Installation of Night Image Vision System (NVIS) / Night Vision Goggles (NVD) Cockpit lighting Validation of FAA STC SR09480RC</td>
<td>JAR27 1st amendment The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AERO DYNAMIX, Inc. Master drawing list MDL 1775-01, Revision M, dated Dec 21st, 2009, or later approved revisions (**Aero DYNAMIX, Inc. Instructions for continued airworthiness (ICA), document No.: ICA MD600-01, Revision N, dated 22 December 2009, or later accepted revisions **)</td>
<td>Refer to applicable RFB for list of equipment mandatory for Night Vision Image System certification</td>
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<td>10028392</td>
<td>EJM AEROSPACE SERVICES</td>
<td>FAA STC ST02531AT</td>
<td>B747-400</td>
<td>Installation of a Matsushita Avionics Systems Corporation System 3000 IFEMAS Flight Entertainment System on a Singapore Airlines Boeing 747-400 aircraft in accordance with Master Drawing List MDL-10270, Rev. 0, dated 06 June 2003 or later FAA approved revision Validation of FAA STC ST02531AT</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>MAS Document No. 14526607 Rev E Instructions for Continued Airworthiness MAS Corp System 3000 IFEMAS Document No. 14526603 Rev. IR Airplane Flight Manual Supplement for MAS Corp System 3000 IFEMAS later revisions of the above listed documents approved by EASA in accordance with EASA Decision 2004...</td>
<td>Not Applicable</td>
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<tr>
<td>10028394</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>REV. 1</td>
<td>PAGGIO P-180 AVANTI II</td>
<td>B747-400</td>
<td>Upgrade of TCAS-4000 to Software Change 7-1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Master Document List (MDL)281F93887-0109 Rev. 1, and Installation Manual 281F93887-1001 Rev. 1 and Instructions for Continued Airworthiness (ICA), ref 281F93887-0701 Rev. 0**</td>
</tr>
<tr>
<td>10028415</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>REV. 1</td>
<td>A320-232</td>
<td>In-Flight Reconfiguration and In-Flight Entertainment (IFE) Installation on Embraer ERJ-190-200LR Revision A320-232 Aircraft Revision of the EASA 10029415 to remove EWIS limitation</td>
<td>In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CFI H.01 (Special Condition H.01) Enhanced Airworthiness Programmes for Aeronautical Systems - ICA on E1WIS The...</td>
<td>- NAT Certification Summary, ref. doc. N0340-D90, Revision 0, dated 31st May 2010 - NAT Master Drawing List, ref. doc. N0340-D10, Revision E, dated 1st April 2010 - NAT EAPAS/FTS, ICA document, ref. doc. N0340-D10, Revision 1, dated 2nd June 2010</td>
<td>None</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>Cert No.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Document</td>
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<td>10028416</td>
<td>BOEING B767-300</td>
<td>REV. 2</td>
<td>TCCA STC</td>
<td>SA04-113</td>
<td>Modification to install the Fwd Pocket door and Mid Pocket door on aseparate circuit installation, Complete Custom Aircraft Interior - Revision to EASA Supplemental Type Certificate 10028416 to incorporate missing drawdown on the MDL - ASIN 9329</td>
<td>EASA A.M.A.205 The Certification Basis for the original product remains applicable to this certificate/approval The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval</td>
<td>-EASA STC No. 10028415 Rev. 1 dated 3 February 2010 - Transport Canada SoC reference C-10-0158 dated 17 May 2010 - TCCA STC No. SA04-113 issue 1 dated 28 October 2004 - Aero Consulting Services Configuration Definition List D864/00 Revision EO dated 09 April 2010 - Instructions for Continued Airworthiness...</td>
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<tr>
<td>10028420</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>FAA STC</td>
<td>ST10963SC</td>
<td>Installation of provisions for an Executive Power System</td>
<td>Certification Basis as per EASA TCDS A.064 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval - CRI H-21 Special Condition H-01 &quot;Enhanced Airworthiness Programme for A318/A319&quot; - CRI A-001 &quot;Executive Airworthiness Programme for A318/A319&quot;</td>
<td>- FAA Supplemental Type Certificate ST10963SC Master Data List, ref. doc. 1110000100, Revision D, dated 17th August 2010 - Airplane Flight Manual Supplement, ref. doc 1112021000, Revision IR, dated 21st December 2009 - ICA Source Document, ref. doc. 1110050099, Rev. IR dated 24th May 2010 later revs.</td>
</tr>
<tr>
<td>10028421</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>FAA STC</td>
<td>ST10963SC</td>
<td>Installation of a Miltope Wireless Access Point</td>
<td>Certification Basis as per EASA TCDS A.064 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval - CRI H-21 Special Condition H-01 &quot;Enhanced Airworthiness Programme for A318/A319&quot;</td>
<td>- FAA Supplemental Type Certificate ST10963SC Master Data List, ref. doc. 1110000000, Revision L, dated 13th July 2010 - Airplane Flight Manual Supplement, ref. doc. 1111021000, Revision IR, dated 16th December 2009 - ICA Source Document, ref. doc. 1110050099, Rev. IR dated 20th May 2010 later revs.</td>
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<tr>
<td>10028425</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 11</td>
<td></td>
<td></td>
<td>Installation of an executive style interior on Airbus A318/A319 aircraft</td>
<td>The Type and OS Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s) CRI H-01 &quot;Enhanced Airworthiness Programme for A318/A319 Systems - ICA on EAWS CRI D-5302-007 'Isolated Compartment(s)' CRI D-5310-007 'GL'&quot;</td>
<td>- Master Data List, ref. doc. 1110000000, Revision W, dated 23rd August 2013 - Airplane Flight Manual Supplement, ref. doc. 1235502159, Revision IR, dated 14th January 2014 - ICA On EWIS Interconnection Systems (EWIS ICA Source Document) ref. doc. 1268008008, Rev.B dated 14 June 2013 - Instructions for Continued...</td>
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<td>10028430</td>
<td>ISRAELI AEROSPACE INDUSTRIES, Ltd.</td>
<td>REV. 6</td>
<td>CAAC</td>
<td>S2Z18</td>
<td>Conversion of Boeing 767-300 from passenger configuration to SpecialFreighter (BDSF) with main deck class E cargo compartment, including aircraft equipped with APB systems Rev. 1: EWIS Instructions for Continued Airworthiness included. AMOC Approved Rev. 2: AFM Supplement updated to Rev 1 and STC Dat...</td>
<td>As per STC Datasheet 10028430. In particular: for Significantly Changed Areas, Certification Basis is CS25 Amendment 2, including - Special Condition(s) CRI D-048 &quot;Configurations -001 through -009 Revolution 11: Conversion of Airbus A318/A319 aircraft per Certification Basis as per EASA TCDS A.064 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval - CRI H-21 Special Condition H-01 &quot;Enhanced Airworthiness Programme for A318/A319&quot; - CRI A-001 &quot;Executive Airworthiness Programme for A318/A319&quot; - CRI D-048 &quot;Configurations -001 through -009 Revolution 11: Conversion of Airbus A318/A319 aircraft per Certification Basis as per EASA TCDS A.064 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval - CRI H-21 Special Condition H-01 &quot;Enhanced Airworthiness Programme for A318/A319&quot; - CRI A-001 &quot;Executive Airworthiness Programme for A318/A319&quot;</td>
<td>An approved Cargo Loading System must be installed in accordance with the STC constitutes an AMOC for the list of affected ADS as published in the STC Datasheet 10028430 Mod Nr. M0234 (for standard Forward Lower Cargo Door) is only applicable to aircrafts equipped with APB...</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10028433</td>
<td>TELAIR US LLC</td>
<td>REV. 1</td>
<td>FAA STC STO2767CH</td>
<td>B767-300</td>
<td>Install Main Deck Cargo Loading System.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this STC Certificate:</td>
<td>Drawing List Report Number 3256, Revision C, 30/09/2008. Applicable revisions or later revisions of the above list of documents approved by EASA in accordance with EASA ED 2004/04/CF (or subsequent revisions of this decision).</td>
</tr>
<tr>
<td>10028446</td>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>REV. 1</td>
<td>BEECH B200GT BEECH B200C BEECH B200CGT BEECH 200 BEECH 2020</td>
<td>Installation of Blue Sky Network (BSN) D1005(A) Iridium Satellite Communication and Tracking</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, B N52 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable.</td>
<td>Installation in accordance with the data listed in Section 4.1 of IATA Alpha AS Master Data List Numbered: MSL EO-B200-23-132 Rev: 07. Operation in accordance with Air Alpha AS Flight Manual Supplement Numbered: FMS EO-B200-23-132 Rev: 00. Instructions for Continued Airworthiness (i.e., Air Alpha AS...</td>
<td>The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved documentation. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other pre...</td>
</tr>
<tr>
<td>10028448</td>
<td>DV AVIATION, Inc. DBA NORTHPOINT AVIATION</td>
<td>REV. 3</td>
<td>FAA STC SAA01B8SE</td>
<td>206H, T206H</td>
<td>Installation of a revised shoulder harness restraint system and new rearmainheadliner in Textron 206H/T206H Rev. 3. Transfer from &quot;Fly-Ten-Yankwe, LLC&quot; to &quot;DV Aviation, Inc.&quot;</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval.</td>
<td>Soloy, LLC Master Drawing List 912-103, dated 31 May 2008. Service Instructions No. 912-01, dated 24 May 2008. Prior to installation of this modification it must be determined that the interrelationship between this design change and any other previously installed design change or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10028449</td>
<td>DV AVIATION, Inc. DBA NORTHPOINT AVIATION</td>
<td>REV. 3</td>
<td>FAA STC SAA080BSE</td>
<td>U206F, T206C, U206G U206A, T206G, 206H, T206H U206A, T206C, 206G, T206D, T206E, U206, T206F U206D, T206E, U206, T206F, T206G, T206B, U206F, T206C, U206G, T206D, U206E, T206B 206, U206B, U206C, T206D 206H, T206H, T206A, U206E,</td>
<td>Install Observation Window Rev. 3. Transfer from &quot;Fly-Ten-Yankwe, LLC&quot; to &quot;DV Aviation, Inc.&quot;</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval.</td>
<td>Soloy, LLC Master Drawing List 901-105, Rev. B, dated 15 November 2000. Manufacturing Inspection Order for Part 961-1101, dated 6 December 2000. This modification has been determined to be compatible with FAA STC SA2153NM and airplanes in landplane, floatplane and skiplane configuration. When modified in accordance with FAA approved data Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10028452</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td></td>
<td>EC 120 B</td>
<td>Replacement of existing SAFT Ni-Cad battery with Concorde lead acid battery</td>
<td>STC ref.</td>
<td>Models, Description, Certification basis</td>
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<tr>
<td>10028464</td>
<td>ANORA INTERNATIONAL LLC</td>
<td>FAA STC</td>
<td>100279LA</td>
<td>B767-300</td>
<td>Install Main Deck Cargo Loading System</td>
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<td>10028465</td>
<td>AERO CONSULTING SERVICES LTD.</td>
<td>TCCA LSTC</td>
<td>C-LSA08-0071D</td>
<td>LEARJET 60</td>
<td>Installation of Chelton HSD Satcom System.</td>
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<td>10028486</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GMBH</td>
<td>SEE ATTACHED MODEL LIST</td>
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<td>Installation of an Aspen EFD-1000 ProSystem.</td>
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<td>10028488</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>GULFSTREAM 150</td>
<td>G150 Steep Approach Landing</td>
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<td>Rev. level</td>
<td>Foreign/NAAC STC no.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10028491</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td></td>
<td>TCCA STC C-LSA09-0110</td>
<td>BD-700-1A10</td>
<td>Bandit Mask Paint Scheme – ASN 9332.</td>
<td>EASA IM.A.008 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: EASA VII/000-050/01, Issue 1 dated 12 January 2010.</td>
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<tr>
<td>10028492</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td></td>
<td>TCCA STC SA04-113</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9332.</td>
<td>EASA IM.A.008 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<tr>
<td>10028493</td>
<td>APICAL INDUSTRIES, Inc.</td>
<td></td>
<td>FAA STC SR00644LA</td>
<td>AS355F2, AS355N, AS355NP AS355E, AS355F, AS355F1</td>
<td>Emergency Float Kit</td>
<td>EASA IM.A.008 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<tr>
<td>10028498</td>
<td>KEYSTONE HELICOPTER CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SR01127NY-D</td>
<td>S-76A, S-76B, S-76C</td>
<td>Installation of a Keystone Medical Interior - See Conditions</td>
<td>EASA IM.A.008 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<tr>
<td>10028501</td>
<td>KEYSTONE HELICOPTER CORPORATION</td>
<td></td>
<td>FAA STC SR00910NY</td>
<td>S-76A, S-76B, S-76C</td>
<td>Structural provisions for the installation of a Spectrolab SX5 Searchlight to the left or right side of the S76.</td>
<td>EASA IM.A.008 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
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<tr>
<td>10028503</td>
<td>KEYSTONE HELICOPTER CORPORATION</td>
<td>FAA STC SR08942NY</td>
<td>S-76A, S-76B, S-76C</td>
<td>Installation of a Liquid Oxygen System for Medic evac use</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Modification/Definition</td>
<td>Keep the installation drawings List IDL00114 dated April 3rd 1997 - Flight Manual Supplement (For the S76A, S76B or S76C) - RFMS No. 00114 dated April 3rd 1997 or later revisions of these documents approved by EASA in accordance with EASA CS 21/10 and Appendix A to the below</td>
<td>- This approval should not be incorporated in any aircraft of this specific model on which other approved modifications are incorporated unless it is determined that there is no adverse effect...</td>
<td>18/02/2010</td>
<td>Active</td>
<td>10028503</td>
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<tr>
<td>10028505</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC NO. ST03746AT-D</td>
<td>GV-SP</td>
<td>Installation of an Airplane Interior for Aircraft Serial 5242.</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Master Indecet.doc, GC514915242 Rev C, dated December 17, 2009, or later approved revisions. To be operated in accordance with FAA approved Aircraft Flight Manual Supplement, ref. GC514915008. Rev. NC, dated December 17, 2009 or later approved rev...</td>
<td>Aircraft Serial Number 5242 only.</td>
<td>18/01/2010</td>
<td>Active</td>
<td>10028505</td>
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<tr>
<td>10028509</td>
<td>CARSON HELICOPTERS, INC.</td>
<td>FAA STC SR02486NY</td>
<td>SIORSKY S61N AND S61NM</td>
<td>Increased Category A Performance.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: FAR 29 at Am 35. Modification/Definition</td>
<td>Carson Helicopters Inc. Certification Plan for S61N increased Category A Performance when equipped with Carson Composite Main Blades - Ref. Report CHI-04-1000Rev B dated 08/23/2006 and Manual Supplement Carson Helicopters Inc RFMS No. 7 Rev 1 dated Jan 27 2009 or later revisions...</td>
<td>As specified in the approved RFMS.</td>
<td>18/01/2010</td>
<td>Active</td>
<td>10028509</td>
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<tr>
<td>10028519</td>
<td>BOMBAUDIER INC.</td>
<td>TCCA STC NO. SA04-113</td>
<td>BD-700-1A10</td>
<td>Installation: Complete Custom Aircraft Interior - ASN 9333</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval - The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval - Transport Canada SoC reference C-09-0993 dated 14 January 2010, TCCA STC No. SA04-113 issue 1 dated 28 October 2004, Aeronautical Services Configuration Definition List D384000 Rev Jamaican dated 21 December 2009, instructions for Continued Airworthiness GC514915242 Rev C, dated December 17, 2009 or later revisions of the above...</td>
<td>Restricted to ASN 9333 and Refer to the Limitations sections of the required AFX systems.</td>
<td>21/01/2010</td>
<td>Active</td>
<td>10028519</td>
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<tr>
<td>10028521</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1 UK STC NO. 00230</td>
<td>BD-700-1A10</td>
<td>Customized Aircraft Completion- ASN 9251 Rev. 1 - Removal of Oxygen Altitude AFMS Limitation (S21-35-96-1010)</td>
<td>The Type and OBD Certification Basis (COB) for the original product remains applicable to this certificate/approval - The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval - STC Twenty One Limited Certification Programme, Doc Ref S21 PROG-1708, Revision 2, dated 30 October 2023; STC Twenty One Limited Master Document List, Doc Ref S21 MDL-1708 Issue 1. dated 30 October 2023 STC Twenty One Limited Flight Manual Supplement, Doc Ref S21 FMS-0057, Issue 2, dated 30 Oct...</td>
<td>Applicable to BD-700-1A10 with Design change S21-25-06-1387, installed under EASA STC 10028521 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>12/03/2024</td>
<td>Active</td>
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<th>STC Holder</th>
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<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>BOMBARDIER INC.</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9115</td>
<td>EASA IMA.009; The Certification Basis for the original product remains applicable to this certificate / approval. Non-significant. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate / approval</td>
<td>Transport Canada STC No. SA04-113 dated 16 January 2010. TCACST No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List DFS0400 Revision DG dated 29 July 2009. Instructions for Continued Airworthiness GC9115000-IFCA. Or later revisions of the above.</td>
<td>Restricted to ASN 9115 and Refer to the Limitations sections of the required AFM Supplement and the referenced ATCM 2-WLAN system is applicable only to IEEE 802.11(b).</td>
<td>21/02/2010</td>
<td>Active</td>
<td>10028553</td>
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<tr>
<td>DASSAULT FALCON SERVICE</td>
<td>Mode II dual transponder installation with elementary and enhanced surveillance capabilities</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate / approval. Compliance has been shown in accordance with the requirements laid down by JAA Temporary Guidance Leaflet JAA Leaflet TLG 13 Certification of...</td>
<td>- DFS Change Approval Sheet DFS 9550-CAS Issue A, dated 20th January 2010. DFS Modification Sheet DFS 9550 Issue A, dated 20th January 2010. DFS Aeroplane Flight Manual Supplement Number DFS F900-030, Original Issue, dated 18th January 2010 or later revisions of the above listed documents approved by E...</td>
<td>This EASA approval is applicable on those Basic Mystere Falcon 900 aircraft previously equipped with - Dual ATC Transponders Honeywell MST-67A PN 066-0143-0601 with 3-axis diversity installation - 2 IRS and 2 ADC AZ 810As per DFS Change Approval Sheet DFS 9550-CAS Issue A, dated 20th January 2010 or...</td>
<td>22/01/2010</td>
<td>Active</td>
<td>10028563</td>
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<tr>
<td>BOMBARDIER INC.</td>
<td>Installation, Interior Completions Improvement Modifications - ASN 9334</td>
<td>EASA TCDS No IM.A.009; The Certification Basis for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>TCCA STC No SA05-60 dated 14th June 2005. Aero Consulting Services Ltd. Configuration Definition List DFS0500 Revision EW dated 22 June 2011. Bombardier Airplane Flight Manual Supplement G-FM25004501 Revision No/Cor later approved revision GC93340000-IFCA. Instructions for Continued Airworthiness...</td>
<td>Restricted to ASN 9334</td>
<td>08/08/2011</td>
<td>Active</td>
<td>10028579</td>
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<tr>
<td>AEROSPACE DESIGN FACILITIES Ltd.</td>
<td>Installation, Interior Completions Improvement Modifications - ASN 9334</td>
<td>EASA TCDS No IM.A.009; The Certification Basis for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate / approval</td>
<td>TCCA STC No SA05-60 dated 14th June 2005. Aero Consulting Services Ltd. Configuration Definition List DFS0500 Revision EW dated 22 June 2011. Bombardier Airplane Flight Manual Supplement G-FM25004501 Revision No/Cor later approved revision GC93340000-IFCA. Instructions for Continued Airworthiness...</td>
<td>None</td>
<td>14/09/2010</td>
<td>Active</td>
<td>10028585</td>
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<td>WESTERN AVIONICS</td>
<td>Installation, Interior Completions Improvement Modifications - ASN 9334</td>
<td>EASA TCDS No IM.A.009; The Certification Basis for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate / approval</td>
<td>TCCA STC No SA05-60 dated 14th June 2005. Aero Consulting Services Ltd. Configuration Definition List DFS0500 Revision EW dated 22 June 2011. Bombardier Airplane Flight Manual Supplement G-FM25004501 Revision No/Cor later approved revision GC93340000-IFCA. Instructions for Continued Airworthiness...</td>
<td>None</td>
<td>27/01/2010</td>
<td>Active</td>
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### Notes
- **STC Holder**: The company or individual responsible for the STC持有者
- **Description**: A brief description of the STC's purpose or function.
- **Certification basis**: The certification basis for the STC, including any applicable standards or guidelines.
- **Associated Technical Documentation**: Information about the associated technical documentation.
- **Limitations/Conditions**: Conditions or limitations associated with the STC.
- **Date of Issue**: The date the STC was issued.
- **Status**: The current status of the STC.
- **System Record**: The record number for the STC system.
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<tr>
<th>STC Holder</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>ADVANCED AVIONICS TECHNOLOGIES/ASM (EASA:IM.A.S.03223)</td>
<td>Installation of dual Collins Mode S Transponders FAA STC ST09365SC, IR, August 14, 1997</td>
<td>FAR 25 as per EASA M-191, Issue 4 (incl. SCh-01 ICA on EWIS)</td>
<td>The Certification Basis for the original product remains applicable folks certificate/approval. The requirements for environmental protection and the associated/credited noise and/or emissions levels of the original product unchanged...</td>
<td>None</td>
<td>19/07/2011</td>
<td>Active</td>
<td>10028594</td>
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<td>ADVANCED AVIONICS TECHNOLOGIES/ASM (EASA:IM.A.S.03220)</td>
<td>Installation of Dual Universal Navigation UNS-1C Flight Management System FAA STC ST09326SC, Rev. 1, dated 14 November 2000</td>
<td>FAR 25 as per EASA M-191, Issue 4 (incl. SCh-01 ICA on EWIS)</td>
<td>The Certification Basis for the original product remains applicable folks certificate/approval. The requirements for environmental protection and the associated/credited noise and/or emissions levels of the original product unchanged...</td>
<td>None</td>
<td>20/07/2011</td>
<td>Active</td>
<td>10028595</td>
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<tr>
<td>AIRLIFT A/S</td>
<td>Installation of Latitude Skynode Satcom and Tracking System</td>
<td>CS 29, Amendment 2, with revision to FAR 29.561 (b)(3) at amendment 02-16</td>
<td>The certification noise and/or emissions levels of the original product unchanged and remain applicable this certificate/approval</td>
<td>None</td>
<td>28/01/2010</td>
<td>Active</td>
<td>10028603</td>
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<tr>
<td>STG AEROSPACE Inc. AKA SAITOLO, Inc. (EASA:IM.A.S.03118)</td>
<td>Installation of STG Aerospace PowerSmart Emergency Primary Power System (WEPPS).</td>
<td>The Certification Basis for the original product remains applicable folks certificate/approval. The requirements for environmental protection and the associated/credited noise and/or emissions levels of the original product unchanged and remain applicable to this approval</td>
<td>Master Drawing List 71670000 rev. D dated 29 October 2009/Construction for Continued Airmobility 71676006 rev. D dated 29 October 2009/Installation of Dual Universal Navigation UNS-1C Flight Management System FAA STC ST09326SC, Rev. 1, dated 14 November 2000</td>
<td>None</td>
<td>03/03/2011</td>
<td>Active</td>
<td>10028633</td>
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<tr>
<td>AERONAUTICAL ACCESSORIES, INC</td>
<td>Installation of AR Passenger Seat Shoulder Harness/Lap Belt Assembly Bell Helicopter Textron 206 B / B Models</td>
<td>AAI Report Number AAI-8590, Revision H, dated 8-21-06</td>
<td>Drawing List/Passenger Seat Shoulder Harness/Lap Belt Assembly for Bell Helicopter Textron 206 B / B Models</td>
<td>None</td>
<td>29/01/2010</td>
<td>Active</td>
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<td>STC Holder</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV 1</td>
<td>FAA STC ST0378AT-D</td>
<td>G150</td>
<td>Installation of an International Communications Group (ICG) Inc NotMail Server</td>
<td>JAR 25, Change 15</td>
<td>The certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/ or emissions levels of the original product unchanged and remain applicable to this certificate/ approval.</td>
<td>-Index ULM GA34428A000, Rev. B, dated 05 November 2009-APMA GA3142884A001, Initial Release, dated 05 November 2009- Instructions for continued airworthiness GA34428A000, Initial Release, dated 05 October 2009 as later revisions of the above listed documents approved by EASA in accordance with EASA ED D ...</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV 1</td>
<td>B200, B200C, B200GT, B200CGT</td>
<td>Alpha STC Definition Document EO-200-23-138 for Intercom installation and modification. Intercom installation in cabin at special mission Aircraft. The cabin intercom interface to the cockp...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Air Alpha Master Data List No. M015-023-1356 later revisions of the above listed documents approved by EASA</td>
<td>The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the approved appropriate documentation. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other pre...</td>
<td>15/06/2013</td>
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<td>TRUATLANTIC MPO, LLC</td>
<td>REV 2</td>
<td>FAA STC ST0635AT</td>
<td>35, 35A, 36, 36A</td>
<td>Deactivation of Aerocore (Learjet) Thrust Reversers/blow the evacuation of the Aerocore (Learjet) Thrust Reversers in accordance with Atlantic Aero Master Drawing List 21193001, rev E, dated 08 February 2003 where STC is a transfer from Atlantic Aero, Greensboro USA, to TruAtlantic Mfg., Kernersville USA...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Airplane Flight Manual Supplement 2119310, rev. IR (revisions approved by EASA ED Decision 2001/401/CE (or subsequent revisions of this decision) and/or the Technical Implementation Pro...</td>
<td>Not Applicable</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td>A318/A319/A320/A321</td>
<td></td>
<td>Approval of Installation Provisions for the use of EFB Class 2 on Airbus A318/A319/A320/A321</td>
<td></td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV 1</td>
<td>B737-300</td>
<td>Zonal Dryer System installation in accordance with 737-EB21-0212</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Supplemental Type Definition Document STO21-2010144M tas. 02Engineering Bulletin 737-EB21-2212 Rev.13br later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>20/10/2010</td>
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<tr>
<td>FORCED AERONAUTIC TECHNOLOGIES</td>
<td>FAA STC SE10926SC</td>
<td>IO-550-N</td>
<td>Installation of BAR Air, LLC supercharger system on Teledyne Continental/Alliances IO-550-N Engines in accordance with Master Drawing ListMDL-6222-1-15-A Revision A, dated April 4, 2008 further FAA approved revision - EASA Validation of FAA STC SE10926SC</td>
<td>JAR-E Change 9 dated May 4, 1997The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Instructions for continued airworthiness are contained in Document No. 22007-30, Revision R01 dated December 1, 2007 prior to later revisions of the above listed documents approved by EASA</td>
<td>The supercharged IO-550-N engine is limited to 29.6 in Hg (204 kPa) MAP/2700 RPM or 310-HP (231 kW)</td>
<td>03/02/2010</td>
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<tr>
<td>10028706</td>
<td>ECMS AVIATION SYSTEMS GMBH</td>
<td>BELL 212, 412</td>
<td></td>
<td>Load Attachment Device BRD01</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approve[EASA] CS 29 Amdt.[2] the certified noise and/or emissions levels of the product unchanged and remain applicable to this certificate/approve.</td>
<td>Flight Manual Supplement for External Load Device BRD01 on Bell 212 and Bell 412, Doc. No. BRD01, FMS_EASA R0, dated 01.12.2006 or later revisions of the above listed documents approved by [EASA].</td>
<td>None</td>
<td>04/02/2010</td>
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<tr>
<td>10028717</td>
<td>AIR COMM CORPORATION LLC</td>
<td>FAA STC SR00893DE</td>
<td></td>
<td>Installation of Cabin Air Conditioning System Provisions</td>
<td>- Master Data List MDD-20083764 REV M-Flight Manual Supplement FMS-20121216 REV F-Flight Manual Supplement FMS-20037664 REV Or later revisions of the above listed document(s) approved/accept under the EASA system.</td>
<td>None</td>
<td>04/02/2010</td>
<td>Active</td>
<td>10028717</td>
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<tr>
<td>10028730</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA LSTC C-4,AIR-338/ D CL-600-2B16</td>
<td></td>
<td>Installation of Complete Custom Aircraft Interior - ASN 5506</td>
<td>FAA STC M-023The Certification Basis for the original product remains applicable to this certificate/approve. The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approve.</td>
<td>TCCA LSTC No. C- LSA09-338D Issue 1 dated 19 December 2008 -Aero Consulting Services Ltd. Modification Data Summary F962600Revision NC or later approved revision - Transport Canada approved Aero Consulting Services Ltd. Airplane Flight/Manual Supplement C7300 Revision A or later approved revision...</td>
<td>None</td>
<td>05/02/2010</td>
<td>Active</td>
<td>10028730</td>
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<tr>
<td>10028747</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>737 ALL MODELS 737-400, 737-500, 737-600 737-700, 737-800 737-900, 737-900ER 737-100, 737-200</td>
<td></td>
<td>Blue Sky Network Satcom installation. This STC installs Satcom communications at the captain and F/O station. Rev. 2: Transfer of STC</td>
<td>The Type and OSD Certification Basis for the original product remains applicable to this certificate/approve. The certified noise and/or emissions levels of the product unchanged and remain applicable to this certificate/approve.</td>
<td>Modification Summary Sheet, ref: MOD-16159-0001 Issue 2 - Master Drawing List, ref: MDL-16192-0001 Issue 2 - Aircraft Flight Manual Supplement, ref: 16119- AFM-0001 Issue 1 for later revisions of the above listed document(s) approved/accept under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously/installed change(s) and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
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<tr>
<td>10028758</td>
<td>CERTWORKS, LLC</td>
<td>FAA STC SR008055E AS560BA AS 350 BA</td>
<td></td>
<td>Installation of Honeywell LTS101-2049A-3A engine on AS 350 BA-Rev 1. Transfer of STC to Cert Works, LLC.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approve. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product unchanged and remain applicable to this certificate/approve without any in...</td>
<td>Modifications completed prior to January 31, 2007 Installation of instructions EHCDA-198-B, Revision 1, dated February 16, 2005.Maintenance Manual Supplement EHCDA-198-B, Revision 2, dated February 18, 2006+ (or) Modifications completed after January 31, 2007 Master Drawing List 920-109, Revision NC, date...</td>
<td>See attached Annex Issue 01 dated 27/11/2020 Prior to installation of change/repair it must be determined that the interrelationship between this change/repair and any other previously/installed change(s) and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/11/2020</td>
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<td>STC Holder</td>
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<td>Foreign/NA/STC ref.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>BAE AEROSPACE LIMITED</td>
<td>REV. 1</td>
<td>TC A1 STC - LSA09-0100</td>
<td>CL-600-2B19</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. C- LSA09-0100 Issue 2 dated 22 July 2009 Aero Consulting Services Ltd. Modification Data Summary G115500 Revision B, Savannah Air Center Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior, G115509 Revision A, and Savannah Air Center Inc. Airplane Flight Manual SAFETYPLANE</td>
<td>Restricted to ASN 8083.</td>
<td>16/08/2013</td>
<td>Active</td>
<td>10028761</td>
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<tr>
<td>ACS-NAI Ltd.</td>
<td>REV. 1</td>
<td>TC A1 STC NO. C- LSA09-0153</td>
<td>BOMBARDEIR CL-600-2B19</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. C- LSA09-0153 Issue 1 dated 02 July 2009 Aero Consulting Services Ltd. Modification Data Summary G098800 Revision D, McRostie Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, G3-269K007 Rev NC Aero Consulting Services Ltd.</td>
<td>Restricted to ASN 8065. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/08/2013</td>
<td>Active</td>
<td>10028762</td>
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<tr>
<td>ACS-NAI Ltd.</td>
<td>REV. 1</td>
<td>TC A1 STC NO. C- LSA09-0153</td>
<td>BOMBARDEIR CL-600-2B19</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. C- LSA09-0153 Issue 2 dated 21 July 2009 Aero Consulting Services Ltd. Modification Data Summary G115600 Revision D, McRostie Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, G3-269K007 Rev NC Aero Consulting Services Ltd.</td>
<td>Restricted to ASN 8078. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/08/2013</td>
<td>Active</td>
<td>10028763</td>
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<tr>
<td>ROEDER PRAESSION GMBH</td>
<td>REV. 1</td>
<td>210</td>
<td></td>
<td>Post installation Approval and conformity check of avionic system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement: C210-NY-07136-08, Rev. 1, Instructions for Continued Airworthiness: C210-NY-07136-07, Rev. 0, Installation Manual: C210-NY-07136-05, Rev. 1, later revisions of the above listed documents approved by EASA.</td>
<td>Limited to S.N 57339. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10028781</td>
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<tr>
<td>SAFRAN ELECTRONICS &amp; DEFENSESUISINE SDI NTT SOLUTIONS &amp; SERVICES</td>
<td>REV. 6</td>
<td>AS 350 BB AS 350 B, AS 350 B1, AS 350 B2</td>
<td></td>
<td>Installation of flight data monitoring and transmitting system SAFETYPLANE or HELICOM V2 ANA or HELICOM V2+ ANA. The original version of this STC approves the installation of the SAFETYPLANE equipment. Revision 1 of this STC incorporates the latest evolutions of this system, identified as HELICOM V2 A...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for: Special Condition(s): CRI F-01 Lithium Battery - The following paragraph (s) at a later amendment: CS 27 initial issue 27.25, 27.27, 27.301, 27.303, 27.305, 27.307.2...</td>
<td>Rotorcraft Flight Manual Supplement for HELICOM V2+ ANA and HELICOM V2+ANA, doc. ref. N1- AS350 B.BA B1 B2, E- C-03, Rev. 7 dated 13/06/2017. Reference Documents doc. ref. N1- AS350 B.BA B1 B2 E- C-08 dated 13/06/2017 or later versions of the above listed document(s) approved/as accepted under the EASA sys...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/02/2020</td>
<td>Active</td>
<td>10028789</td>
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<td>STC Holder</td>
<td>FAA STC</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AERONAUTICAL ACCESSORIES, INC</td>
<td>SR02160LA</td>
<td>BELL 407</td>
<td>Installation of Mounting Provisions for a Tail Boom Radar Altimeter in accordance with Meeker Aviation Drawing List for Radar Altimeter Mount Kit AA-TBA-16-010, and Document List of Radar Altimeter Mount Kit AA-TBA-16-011.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- Meeker Aviation Drawing List for Radar Altimeter Mount Kit AA-TBA-16-010, Revision A, dated August 18, 2008. Document List of Radar Altimeter Mount Kit AA-TBA-16-011.</td>
<td>This STC provides mounting provisions only; it does not allow the holder to install an altimeter or any wiring to power an altimeter.</td>
<td>09/02/2010</td>
<td>Active</td>
<td>10028799</td>
<td></td>
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<tr>
<td>AERONAUTICAL ACCESSORIES, INC</td>
<td>STD1210A</td>
<td>BELL 407</td>
<td>Installation of a Tail Boom Antenna Mount Kit in accordance with Meeker Aviation MDL for VHF Communication Antenna Mount Kit AA-TBA-16-010.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- Meeker Aviation MDL for VHF Communication Antenna Mount Kit AA-TBA-16-010.</td>
<td>This STC provides mounting provisions only; it does not allow the holder to install an antenna or any wiring to power an antenna.</td>
<td>10/02/2010</td>
<td>Active</td>
<td>10028797</td>
<td></td>
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<tr>
<td>AERO CONSULTING SERVICES LTD.</td>
<td>C-SA101S-006D</td>
<td>9003P</td>
<td>Installation of Dual Liftport Patient Loading and Utility System (PLUS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- Installation in accordance with Aero Consulting Services Ltd. Flight Manual Supplement G569000 Rev N/C or later approved revisions - Operation in accordance with Aero Consulting Services Ltd. Flight Manual Supplement G569000 Rev N/C or later approved revisions - Maintenance in accordance with EASA ED.</td>
<td>Aircraft Serial Number HA-0114 Only</td>
<td>10/02/2010</td>
<td>Active</td>
<td>10028798</td>
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<td>STC No.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10028816</td>
<td>AVONIK STRAUSSING ENTWICKLUNGS GmbH</td>
<td></td>
<td></td>
<td>B. M28L, M28K, M28L, B. M28C, M28E, M28F, M28G, A. M28M, M28R</td>
<td>Installation of Aspen EFD1000 PRO System.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to the aircraft identifier approval C5-25, Initial issue (not required by 21A 101, where this is necessary) (the certified noise and/or emissions levels of the original product are unchanged).</td>
<td>None.</td>
<td></td>
<td>11/02/2010</td>
<td>Active</td>
<td>10028816</td>
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<td>10028823</td>
<td>DART AEROSPACE Ltd.</td>
<td></td>
<td></td>
<td>TCCA STC SH06-16</td>
<td>Traffic Advisory System Installation.</td>
<td>The original product certification basis as per EASA TCDS R.009 Issue 03/12, October 2007 The Certification Basis for the original product remains applicable to the certificate/approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this...</td>
<td>None.</td>
<td></td>
<td>12/02/2010</td>
<td>Active</td>
<td>10028823</td>
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<tr>
<td>10028830</td>
<td>LUFTWAJSA TECHNIK AG</td>
<td>REV. 3</td>
<td>A319-133</td>
<td>A319-133 [GAF VIP Cabin Completion] Rev: 1; Closure of post certification action D-08-01Rev: 2; Updated Instructions for Continued AirworthinessRev: 3; Updated Instructions for Continued Airworthiness Basis as amended by the following additional or alternative airworthiness requirements: Special Condition(s): D-4302-002 “Isolated Compartments”; D-4310-002 “Glass Screen of Display/Monitors”; D-4314-002 “Shower Installation Equivalent Safety Find...</td>
<td>Installation of Dual MLG (DIZ32903T &amp; DIZ32903T) and NLG (DIZ32816T).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between the change/repair and any otherwise installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/04/2023</td>
<td>Active</td>
<td>10028830</td>
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<tr>
<td>10028837</td>
<td>DUNLOP AIRCRAFT TYRES LIMITED</td>
<td>REV. 6</td>
<td>CAA UK RESPONSIBILITIES</td>
<td>ERU 190-200STD ERU 190-200LR ERU 190-200GW ERU 190-100STD ERU 190-100LR ERU 190-100GW</td>
<td>DURO Tyres Ltd.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td></td>
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<td>27/06/2016</td>
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<tr>
<td>10028844</td>
<td>AVIATION PARTNERS, INC.</td>
<td>REV. 3</td>
<td>FAA ST01987S</td>
<td>FALCON 2000, FALCON 2000EX</td>
<td>Revision of EASA STC EASA 8M.A.S. 02043 (“TWA STC ST01987S Installation of blended winglet on Dassault Falcon 2000 and 2000EX Series airplanes, including 2000EX EASA designation airplane(s)”) to include Falcon 2000/900 Series and 2000EX Engine Flight Manual (AFM) Revision for Forward Concentration of Gravity (CG) Limitations...</td>
<td>The Type and O Caldwell Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change with an impact on the noise database and/or emissions levels. The environmental protection requirements are amended as follows: C3 96 amendment...</td>
<td></td>
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<td>29/04/2024</td>
<td>Active</td>
<td>10028844</td>
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<tr>
<td>10028847</td>
<td>TENENCIA Ltd.</td>
<td>REV. 2</td>
<td>AVRO 146-RJ100</td>
<td>Auro 146-RJ100 Avro Flight VIP Installation.</td>
<td>EASA TCDS A.182 The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to the aircraft identifier approval CRI-H-01 1 Instructions for Continued Airworthiness (ICA) on Electrical/Wiring Interconnection Systems (EHIS) Issue 2 dated T...</td>
<td>Modification TEN 621, BE2 Systems Avro 146-RJ100 Avro Flight VIP Installation, Issue 5 dated 24 January 2011 later revisions of the above listed documents approved by EASA</td>
<td>Serial number E3387 only.</td>
<td>26/01/2011</td>
<td>Active</td>
<td>10028847</td>
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<td>EASA STC no</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NA</td>
<td>STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10028850</td>
<td>EUROCOPTER UK Ltd.</td>
<td></td>
<td>CAIA UK RESPONSIBILITY</td>
<td>BK117 C-2</td>
<td>STC</td>
<td>STC Holder 10028851 Active Models 16/02/2010 15/11/2010 Active System 10028872 Status Development and Installation of High Visibility Crew Door in accordance with Aeronautical Accessories, Inc. Drawing List, Report No AA-01073</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>TRM, C-101-01572, Aeronautical Accessories, Inc. Drawing List No. 156-144-00, Rev. 0, dated 7/22/2009 and/or revisions of the above listed documents approved by EASA in accordance with EASA-ED Decision 2009-0408/CF or subsequent revisions to this decision</td>
<td>None</td>
<td>16/02/2010</td>
<td>Active</td>
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<tr>
<td>10028851</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>REV. 1</td>
<td>CAIA UK RESPONSIBILITY</td>
<td>MBB-BK117 C-2</td>
<td>Installation of Spectrovia Nightsun XP Skid Mounted System: Equipment inspection and recertification to close the requirement for AD No.2010-0183R1.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Aeronautical Accessories, Inc. Drawing List No. 156-144-00, Rev. 0, dated 7/22/2009 and/or revisions of the above listed documents approved by EASA in accordance with EASA-ED Decision 2009-0408/CF or subsequent revisions to this decision</td>
<td>None</td>
<td>15/11/2010</td>
<td>Active</td>
<td>10228851</td>
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<tr>
<td>10028856</td>
<td>EUROCOPTER UK Ltd.</td>
<td></td>
<td>CAIA UK RESPONSIBILITY</td>
<td>BK117 C-2</td>
<td>Cabin Configuration Definition of permissible configurations and combinations thereof individually approved changes based on EC UK Major Change MHL/145/1229/A into the BK117 C-2.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Aeronautical Accessories, Inc. Drawing List No. 156-144-00, Rev. 0, dated 7/22/2009 and/or revisions of the above listed documents approved by EASA in accordance with EASA-ED Decision 2009-0408/CF or subsequent revisions to this decision</td>
<td>None</td>
<td>16/02/2010</td>
<td>Active</td>
<td>10228856</td>
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<tr>
<td>10028872</td>
<td>AERONAUTICAL ACCESSORIES, INC</td>
<td></td>
<td>FAA STC SR00329RC</td>
<td>BELL 407</td>
<td>Installation of Bulged Skylight in accordance with Aeronautical Accessories, Inc. Drawing List, Report No AA-01070</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Aeronautical Accessories, Inc. Drawing List No. 156-144-00, Rev. 0, dated 7/22/2009 and/or revisions of the above listed documents approved by EASA in accordance with EASA-ED Decision 2009-0408/CF or subsequent revisions to this decision</td>
<td>None</td>
<td>16/02/2010</td>
<td>Active</td>
<td>10228872</td>
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<tr>
<td>10028873</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td></td>
<td>FAA STC SR00208SE</td>
<td>EC 120 B</td>
<td>Cargo hook kit Model 200-351-00</td>
<td>As specified in TCDS DGAC No. 1697 The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Aeronautical Accessories, Inc. Drawing List, Report No AA-01073</td>
<td>None</td>
<td>16/02/2010</td>
<td>Active</td>
<td>10228873</td>
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<td>10028891</td>
<td>AERONAUTICAL ACCESSORIES, INC</td>
<td></td>
<td>FAA STC SR00329RC</td>
<td>BELL 407</td>
<td>Development and Installation of High Visibility Crew Door in accordance with Aeronautical Accessories, Inc. Drawing List, Report No AA-01073</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Aeronautical Accessories, Inc. Drawing List No. 156-144-00, Rev. 0, dated 7/22/2009 and/or revisions of the above listed documents approved by EASA in accordance with EASA-ED Decision 2009-0408/CF or subsequent revisions to this decision</td>
<td>None</td>
<td>17/02/2010</td>
<td>Active</td>
<td>10228891</td>
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<td>STC Holder</td>
<td>STC no</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC no</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>AERONAUTICAL ACCESSORIES, INC</td>
<td>1002893</td>
<td>FAA STD SR09379RC-O</td>
<td>1</td>
<td>BELL 407</td>
<td>High Visibility Door Kits in accordance with Aeronautical Accessories, Inc. Drawing List Report No AA-05155</td>
<td>The Certification Basis for the original product remains applicable. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval</td>
<td>-Drawing List Report No AA-05155, Revision K dated 14th December 2009</td>
<td>None</td>
<td>17/02/2010</td>
<td>Active</td>
<td>10028932</td>
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<tr>
<td>LATITUDE TECHNOLOGIES CORPORATION</td>
<td>10028910</td>
<td>FAA STC ST09375C</td>
<td>1</td>
<td>DHC-8-401, DHC-8-402, DHC-8-201, DHC-8-202</td>
<td>Skynode 5200-xxx SATPHONE/TRAVERSE/ TRACKER system product design in accordance with Latitude Technologies Corporation Skynode 5200-xxx Master Document List No S200-000-001 Revision 1.09, dated Aug 10, 2009.</td>
<td>As defined in FAA TCDS H459V1. The Certification Basis for the original product remains applicable to the revised certificate/approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Latitude Technologies Corporation Skynode 5200-xxx Rotorcraft Flight Manual Supplement No S200-402-002, Revision 01. (EASA specific RFM) Supplement or later revisions of the above listed documents approved by Latitude Technologies Corporation Skynode 5200-xxx Master Document List No S200-000-001</td>
<td>None</td>
<td>18/02/2010</td>
<td>Active</td>
<td>10028910</td>
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<tr>
<td>ADVANCED AVIONICS TECHNOLOGIES INC</td>
<td>10028911</td>
<td>FAA STC ST09375C</td>
<td>1</td>
<td>DHC-8-401, DHC-8-402, DHC-8-201, DHC-8-202</td>
<td>Installation of Coltech ATSCALL tracker system product design in accordance with Latitude Technologies Corporation Skynode 5200-xxx Master Document List No S200-000-001 Revision 1.09, dated Aug 10, 2009.</td>
<td>Updated Certification Basis as defined in FAA TCDS H459V1. The Certification Basis for the original product remains applicable to the revised certificate/approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Installation of Coltech ATSCALL tracker system product design in accordance with Latitude Technologies Corporation Skynode 5200-xxx Master Document List No S200-000-001, Revision 1.09, dated Aug 10, 2009</td>
<td>None</td>
<td>20/07/2011</td>
<td>Active</td>
<td>10028911</td>
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<tr>
<td>DART AEROSPACE Ltd</td>
<td>10028917</td>
<td>FAA STC ST09375C</td>
<td>1</td>
<td>BELL 412EP, BELL 204B, 214B-1, BELL 204B, 205A-1</td>
<td>&quot;Full Vertical Reference Door for Left Hand Vertical Reference/Operations&quot;</td>
<td>As defined on FAA TCDS H459V1. The Certification Basis as defined in FAA TCDS H459V1. The Certification Basis for the original product remains applicable to the revised certificate/approval.</td>
<td>-For DHC-8-201, DHC-8-202</td>
<td>Full compliance has not been demonstrated for airworthiness standards 29.1303 (g)(h)(i)(j), 29.1321(b) and 29.1523(b) for single pilot operation from the Left Seat resulting in a Restricted Category classification.</td>
<td>10/03/2010</td>
<td>Active</td>
<td>10028917</td>
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<tr>
<td>EUROCOPTER UK Ltd</td>
<td>10028925</td>
<td>FAA STC ST09375C</td>
<td>1</td>
<td>BELL 412EP, BELL 204B, 214B-1, BELL 204B, 205A-1</td>
<td>&quot;Full Vertical Reference Door for Left Hand Vertical Reference/Operations&quot;</td>
<td>As defined on FAA TCDS H459V1. The Certification Basis as defined in FAA TCDS H459V1. The Certification Basis for the original product remains applicable to the revised certificate/approval.</td>
<td>-Installation of Coltech ATSCALL tracker system product design in accordance with Latitude Technologies Corporation Skynode 5200-xxx Master Document List No S200-000-001, Revision 1.09, dated Aug 10, 2009.</td>
<td>None</td>
<td>19/02/2010</td>
<td>Active</td>
<td>10028925</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>STO1409Wk-D</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10028927</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC</td>
<td>STO1409Wk-D</td>
<td>MYSTERE FALCON MF900</td>
<td>EASA Validation of FAA STC STS1409Wk-D Installation of an LJ3Communications, On-3100 Electronic StandbyInstrument System (ESIS) in accordance with Duncan Aviation Inc Master/Document List No 00622063 Revision C, dated February 19th, 2008 or later revisions approved by EASA in accordance with EASA ED D...</td>
<td>-Duncan Aviation Master Document List No 00622063, Revision C, dated February 19th, 2008 -Duncan Aviation Aircraft Flight Manual Supplement, No. 00622020, Revision A, originally approved October 19th, 2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED D...</td>
<td>Validation limited to Basic Mystere Falcon MF900 aircraft model</td>
<td>19/02/2010</td>
<td>Active</td>
<td>10028927</td>
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<tr>
<td>10028933</td>
<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td>PA-46-310P</td>
<td>Installation of Garmin GNS530W, GNS430W and GMA347</td>
<td>CS-23</td>
<td>Installation must be in accordance with AirPlus Master Document List: ID-0359-340303, Rev. C, dated 19th Feb. 2010 or later approved revisions. Operation must be in accordance with AirPlus Flight Manual Supplement, PO-0359-34030, PO-0359-340303 and PO-0359-340401 Rev. Dated 03rd Sept. 200...</td>
<td>None</td>
<td>07/02/2021</td>
<td>Active</td>
<td>10028933</td>
<td></td>
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<tr>
<td>10028945</td>
<td>CESSNA DUESSELDORF CITATION CENTER GmbH</td>
<td>REV. 1</td>
<td>P.180 AVANTI, AVANTI II</td>
<td>Installation of In-Flight Entertainment System in accordance with Engineering Order, Doc. No. 0009-9</td>
<td>The Certification Basis for the original product remains applicable to the design change. This certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval</td>
<td>-EASA approved Engineering Order, Doc. No. 0009 Revised, dated 05-Oct-2009 or later approved issue EASA approved Aircraft Flight Manual Supplement, Doc. No. 20099, Revision A, dated 05-Oct-2009, or later approved issue EASA approved Aircraft Flight Manual Supplement, Doc. No. 20099, Revision A, dated 05-Oct-2009, or later approved issue EASA approved Aircraft Flight Manual Supplement, Doc. No. 20099, Revision A, dated 05-Oct-2009, or later approved issue</td>
<td>None</td>
<td>22/02/2010</td>
<td>Active</td>
<td>10028945</td>
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<tr>
<td>10028946</td>
<td>TQAS - TOTAL QUALITY AND AIRCRAFT SOLUTIONS ONS, LDA</td>
<td>REV. 1</td>
<td>B737-500</td>
<td>Installation of Floor proximity emergency photo luminescent path.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the design change and remain applicable to this certification approval</td>
<td>-Design Dossier CG-008-09-CPO dated February 11th, 2010 -AFM Supplement CG-008-09-AFM Iss. 1 dated 03rd Sept. 200...</td>
<td>None</td>
<td>22/02/2010</td>
<td>Active</td>
<td>10028946</td>
<td></td>
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<tr>
<td>10028996</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 2</td>
<td>A319-115</td>
<td>ACJ Centre Modification C20040 - VIP Cabin Interior - &quot;ACROPOLIS&quot; Revision due to update of AFM-S and MPD-S</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements as listed in the General CRI G-411 &quot;ACJ Centre Modification C20040 - VIP Cabin Interior - &quot;ACROPOLIS&quot;&quot;. The requirements for environmental protection and the associated certified noise and/or emissions levels...</td>
<td>1. Modification Approval Sheet ref. C20400, Issue 3 &quot;Top Modifications/ACROPOLIS&quot; dated 24th October 2013. AirPlus Supplement Ref. FM-S MOD C20400, Rev. 02, dated 1st October 2013. MPD Supplement Ref. MPD-S MOD C20400, Rev. 03, dated 1st October 2013 or later revisions of the above listed documents approved by EASA</td>
<td>This STC is limited to Airbus A319 MSN 35285 prior to installation of this design change and may otherwise adversely affect the airworthiness of the product. The...</td>
<td>22/11/2013</td>
<td>Active</td>
<td>10028996</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>STC ref.</td>
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<td>10029002</td>
<td>GAMA ENGINEERING LTD.</td>
<td></td>
<td>CAS 10-0088-08</td>
<td>PA-31</td>
<td>Installation of defence and antennas and systems</td>
<td>CS-23.2(b)(3) The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions are unchanged</td>
<td>None</td>
<td></td>
<td>25/02/2010</td>
<td>Active</td>
<td>10029002</td>
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<tr>
<td>10029004</td>
<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td>CESSNA 501, 551</td>
<td>EACE 23.2(b)(3) The certification noise and/or emissions are unchanged and remain applicable to this certificate/approval</td>
<td>Flight Manual</td>
<td>Supplements PO-0673-340-001, Rev. 1</td>
<td>None</td>
<td></td>
<td>07/02/2021</td>
<td>Active</td>
<td>10029004</td>
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<tr>
<td>10029007</td>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td></td>
<td>EACE 58P</td>
<td>BEECH 258-09 - Radio R-NAV av es couplage PA</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Flight Manual</td>
<td>Document de vol SEAM/MV/2789/09</td>
<td>Limited to: S/N TJ-74</td>
<td>26/02/2010</td>
<td>Active</td>
<td>10029007</td>
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<tr>
<td>10029061</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST03752AT-D</td>
<td>GIV-X</td>
<td>Aircraft completion and installation of an Airplane Interior, S/N 4180</td>
<td>JAR 23 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Flight Manual</td>
<td>Document de vol JAR 23</td>
<td>This approval is limited to S/N 4180 only. The following STC's are required part of this</td>
<td>11/06/2020</td>
<td>Active</td>
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<tr>
<td>10029066</td>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 2</td>
<td>B747-400</td>
<td>B747-25-1005-MDL</td>
<td>Reconconfiguration of Virgin Atlantic Airlines</td>
<td>B747-25-1005-MDL</td>
<td>Instructions for Continued Airworthiness Rev. OC747-25-1005-MDL-01</td>
<td>Approval of the Life Vests is not a part of this STC.</td>
<td>20/04/2021</td>
<td>Active</td>
<td>10029066</td>
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<tr>
<td>10029073</td>
<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td>(EASA.A.6.03062)</td>
<td>B747-200</td>
<td>Video transmission mast under a Beech 200.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Flight Manual</td>
<td></td>
<td>All associated Audio/Video Relay electronic systems and antennas must be removed before VOR navigation and before beginning the approach.</td>
<td>01/03/2010</td>
<td>Active</td>
<td>10029073</td>
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<td>STC Holder</td>
<td>STC STC no.</td>
<td>FA2 STC</td>
<td>Aircraft</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>10029091</td>
<td>REV 1</td>
<td>747-400</td>
<td>Introduction of changes to the cabin layout in accordance with STC Modification Summary MOD-11427-0001/Part A: introduces a seating configuration of 22 business class and 44 economy class PassB: introduces new emergency equipment necessary to comply with the requirements of 22/440Y passengers operating...</td>
<td>The Type and OSd Certification Bases for the original product are amended by the following additional or alternative airworthiness requirements: The certification basis is CS25 at change 4 for changed parts except for the following requirement FAR 25.83(d) at amendment 857 for the requirements for environmental...</td>
<td>EASA AFM supplement 11427-AFM-0001 Issue 1, Installation of Photo-luminescent Aisle Path Marking System, shall be used with the following Instructions for Continuing Airworthiness included in ATC Mod Summary MD011427-0000, Issue 4.</td>
<td>Prior to installation of this change/repair it must be determined if the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
<td>Active</td>
<td>10229001</td>
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<tr>
<td>GAMA ENGINEERING LTD.</td>
<td>10029092</td>
<td>REV 1</td>
<td>PA-31</td>
<td>ATC transponder upgrade for Mode S enhanced surveillance</td>
<td>CS-23 The certification noise and/or emissions levels of the original production unchanged and remain applicable to the certificate/approval</td>
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<td>B/E AEROSPACE, Inc. - FSI</td>
<td>10029094</td>
<td>REV 1</td>
<td>FFA STC ST02561SE</td>
<td>Remove EWIS limitation for EASA STC No. 10029094 - First Class Seat Installation</td>
<td>FAA STC ST02561SE</td>
<td>Concurrent installation of EASA STC No. 10029095 (EASA Project No.EASA.IM.A.S. 03043) is required Concurrent installation of EASA STC No. EASA.IM.A.S.01797 is required</td>
<td>04/10/2010</td>
<td>Active</td>
<td>10260904</td>
<td></td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>10029095</td>
<td></td>
<td>B777-200</td>
<td>First Class Cabin Upgrade - BA 777-200 Aircraft</td>
<td>FAA STC ST02562SE</td>
<td>Concurrent installation of EASA STC No. 10029096 (EASA Project No.EASA.IM.A.S. 02976) is required Concurrent installation of EASA STC No. EASA.IM.A.S.01797 is required</td>
<td>03/02/2010</td>
<td>Active</td>
<td>10229005</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10029101</td>
<td>AEROLITE AG</td>
<td>REV. 5</td>
<td></td>
<td>B200, B200GT</td>
<td>Emergency Medical Services Interior Equipment Kit: P/N 230001-501 and P/N 230001-502.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this STC/certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any...</td>
<td>Installation of EMS equipment in accordance with Aeronautic document MDL-09-009, at revision G, dated February 13, 2020. Operation and Maintenance of Kit P/N 230001-501 in accordance with Aircraft Flight Manual Supplement, Aeronautic document FMS-09-005, at revision B, dated June 25, 2015. Oper...</td>
<td>03/02/2020</td>
<td>Active</td>
<td>10029101</td>
<td></td>
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<tr>
<td>10029103</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>REV. 3</td>
<td></td>
<td></td>
<td>Installation of Runway Turnoff and Logo Lights. Minor Change to EASA STC 10029103, Revision 1: issue date 31.03.2010 granted to H4 Aerospace (UK) Ltd. Minor change approval is sought to reflect the up issue wording of the runqways of continued airworthiness including the FAA approval initial airworthiness...</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval (RAI-H01 / SC Enhanced Airworthiness Programme for Aircraft Systems -ICA on EWIS): The certificated noise and/or emissions levels of the...</td>
<td>Modification Summary Sheet and Declaration of Compliance doc. N &quot;HA1269 Issue 5 dated 10/09/10; Instructions for Continued Airworthiness doc. N &quot;HA4D0512 Issue 3 dated 10/09/2010 and N &quot;HA4D0513 Issue 3 dated 10/09/2010 - Master Document List doc. N &quot;HA4D0525 Issue 5 and N&quot; HA4D0526 Issue 4 - Master.</td>
<td>Restricted to the aircraft listed in doc. N: HA4DR530 - Master/Effective List for Goodrich RTO and Logo Light Installation.</td>
<td>20/10/2010</td>
<td>Active</td>
<td>10029103</td>
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<tr>
<td>10029105</td>
<td>NASCO AIRCRAFT BRAKE, INC.</td>
<td>REV. 1</td>
<td></td>
<td>747-300 SERIES, 747-200F SERIES, 747-200B SERIES, 747-200C SERIES, 747-200F SERIES</td>
<td>Installation and Replacement of Steel Brake Components in accordance with NASCO’s Configuration Control Drawing Report Number NCC 005 Revision A, dated December 2, 2008, or later FAA approved revision.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the...</td>
<td>None</td>
<td>05/03/2010</td>
<td>Active</td>
<td>10029105</td>
<td></td>
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<tr>
<td>10029107</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>Fabrication of DART Aerospace USA, Inc. D119-647-D111/013 Bearpaws and installation of this modification.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later revisions of the FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original prod...</td>
<td>1. Master Document List MDL-D119-647, Revision G, dated 5 March 2020, or later FAA approved revision. 2. Installation Instructions IN-D119-647, Revision B, dated 1 November 2019, or later FAA approved revision. 3. Instructions for Continued Airworthiness: ICAO: D119-647-01; &lt;...</td>
<td>1. Model A109S must be operated in accordance with Rotorcraft Flight Manual Supplement (RFMS) FMS- D109-647. Revision A, dated 8 April 2020, or later FAA approved revision. 2. Model A109S must have Trekker kit part number 10900000F01 installed. 3. Prior to installation of this design change it must be...</td>
<td>14/06/2020</td>
<td>Active</td>
<td>10029107</td>
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<tr>
<td>10029112</td>
<td>CESKA LETECKA SERVISNI A.S.</td>
<td>REV. 5</td>
<td></td>
<td></td>
<td>Dual Mode-S Transponder upgrade for EHS capability</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23, Amend 1, dated 12/02/2009 (Special Conditions: EASA INT POL 231 (HRP) Flight Manual Supplements: No. AC106-2501; FMU Unit UNS-1te, issued January 2010-No. AC106-2502; Supplement No. 5 - TCAS-94 (997.0) a odpovídající s Modu TDR-94D Rockwell Collins, issued January 2010.</td>
<td>None</td>
<td>05/03/2010</td>
<td>Active</td>
<td>10029112</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>Phoenix Aerospace Limited</td>
<td>CAU</td>
<td>AS355F2, AS355N, AS355Np, AE355F, AE355F1</td>
<td>Installation of Skyeye Ltd Traffic avoidance and Honeywell's KMD250 MFD</td>
<td>As specified on EASA TDS R 146The Certification Basis for the original product remains applicable to this certificate/revision approvedCS-27 change 2 date 17 November 2008 used for this STCThe certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/revision</td>
<td>-PAD/STC/EASA/EASA/001 revision A - Flight Manual Supplement</td>
<td>Not Applicable</td>
<td>08/02/2010</td>
<td>Active</td>
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<td>Luftansa Technik AG</td>
<td>REV.</td>
<td>A340-600</td>
<td>A340-600 Cabin Retrofit Ethad ETD08EASA STC 10029134 revision due to removal of EWIS limitation</td>
<td>The Certification Basis for the original product as amendedby the following additional or alternative airworthiness requirements - Special Condition (s): CRI E-19 &quot;Mrs suite&quot;, the following paragraph(s) at a later amendment: CS-25 Am625., Appendix H &quot;Electrical Wiring Interconnection System&quot; The re...</td>
<td>-Supplemental Type Definition Document, ref. Doc. No STD200-09/2006, Issue 1, dated 24th February 2010, Compliance Record List ref. doc. No. CRL 00-06/2006 Iss 2, dated 24th April 2013, Instructions for Continued Airworthiness, ref. Doc. No A-25-10/011-ICA, Issue 1, dated 15th February, 2010, EWISA,-ECU RN6 Number 9, Issue 1or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>13/05/2013</td>
<td>Active</td>
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<td>Eurocopter UK Ltd.</td>
<td>CAU</td>
<td>MBB-BK117-C2</td>
<td>Martin Baker Servicing Passenger Seats</td>
<td>EASA TDS R 010The Certification Basis for the original product remains applicable to this certificate/ approval</td>
<td>-Engineering Order 725EC0280010-01A—TOH Supplement for Cessna P210, S/N P21000209, Doc. No. 7725 AM 280010 01A, Revision 1, dated 18 February 2010, Limited to Aircraft S/N P21000209</td>
<td>None</td>
<td>06/02/2010</td>
<td>Active</td>
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<td>Roeder Premision GmbH</td>
<td></td>
<td>CESSNA P210N</td>
<td>Ferry Tank Provisioning</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval</td>
<td>-Engineering Order 725EC0280010-01A—TOH Supplement for Cessna P210, S/N P21000209, Doc. No. 7725 AM 280010 01A, Revision 1, dated 18 February 2010, Limited to Aircraft S/N P21000209</td>
<td>None</td>
<td>11/03/2010</td>
<td>Active</td>
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<td>DART AEROSPACE Ltd.</td>
<td></td>
<td>TCCA STC SH95-11 ISSUE 5</td>
<td>&quot;Adjustable Engine Mount Kit&quot;</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval</td>
<td>-MOD/205-708 Rev. C- Master Document List IM- D205-708 Rev A Installation Instructions-ICAI-205-708 Rev. 0 Instructions for Continued Airworthiness</td>
<td>None</td>
<td>11/03/2010</td>
<td>Active</td>
<td></td>
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<td>Buchar Leichtbau AG</td>
<td>REV-</td>
<td>EC635 T1, EC635 T2+, EC635 T3, EC635 P2+, EC635 P3, EC135 T2+, EC135 T3 EC135 T1, EC135 T2, EC135 P2, EC135 P3, EC135 P1, EC135 P2</td>
<td>Bucher Ambulance and Rescue System (EMS Installation) A4A-EC135-AC67/01 of revisions to STC 10029211 for EMS Bucher Ambulance &amp; Rescue System P/N A4A-EC135-AC67 Initial Revision- The presented EMS Equipment describes a fully integrated interior orthomedical support devices. See Equipment List EC15-65...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval</td>
<td>- Flight Manual Supplement FMS-AC87/01 Revision 0-01A Master Document List E50012480 Revision NC-Operation Manual Supplement E214-01-20 Revision 024 later revisions of the above listed documents/approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/04/2021</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV 1</td>
<td>FAA STC ST03738AT-D</td>
<td>GIV-X, GIV-SP</td>
<td>Installation of Fwd &amp; Aft Lavatory Smoke Detectors.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>-Installation in accordance with Gulfstream Aerospace Corporation ST03648AT-D, Revision H, dated October 27, 2010 (or later revisions of the above listed documents approved by EASA).</td>
<td>None</td>
<td>10/05/2012</td>
<td>Active</td>
<td>10029213</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST03542AT-D</td>
<td>GULFSTREAM GIV-SP, GIV-X</td>
<td>Installation of a Generation 3, Broad Band Multi-Link (BBML) System.</td>
<td>JAR 25 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>-Installation in accordance with Gulfstream Aerospace Corporation ST03648AT-D, Revision H, dated October 27, 2010 (or later revisions of the above listed documents approved by EASA).</td>
<td>The installer must determine whether this design change is compatible with previously approved modifications.</td>
<td>12/03/2010</td>
<td>Active</td>
<td>10029215</td>
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<tr>
<td>BEECHCRAFT CORPORATION</td>
<td>FAA STC SA15026W-D</td>
<td>B300 SERIES</td>
<td>STC SA15026W-D issued to allow installation of new Lash 6000 and single Lash 6000 units in a Hawker Beechcraft Corporation King Air B300/B300C aircraft.</td>
<td>FAR 23 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>-Installation in accordance with BeechcraftCorporation ST03532AT-D, Revision H, dated September 10, 2009 (or later revisions of the above listed documents approved by EASA).</td>
<td>None.</td>
<td>12/03/2010</td>
<td>Active</td>
<td>10029216</td>
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<tr>
<td>BEECHCRAFT CORPORATION</td>
<td>FAA STC ST03749AT-D</td>
<td>GIV-X</td>
<td>Installation of cabin interior and Airplane Completion for Serial 4176.</td>
<td>JAR 25 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>-Installation in accordance with Master Index List GCA414914176, Revision H, dated January 27, 2010 (or later revisions of the above listed documents approved by EASA).</td>
<td>This approval is limited to only the installation made in Aircraft S/N 4176.</td>
<td>12/03/2010</td>
<td>Active</td>
<td>10029220</td>
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<td>AERO CONSULTING SERVICES LTD.</td>
<td>TCCA 57STC C-LSA10-010/ D</td>
<td>PC-12/47E</td>
<td>Installation, Life/Service PLUS Units in Tandem Configuration</td>
<td>Canadian Airworthiness Manual Chapter 503, Change 3 dated January 2, 1993 (which is equivalent to 14 CFR 23, Normal Category, effective February 4, 1991, including Amendments 23-1 through 23-42). The Certification Basis for the original product is the following additional or alternative airworthiness req...</td>
<td>-Installation in accordance with AOC-NAI Ltd. Modification Data Summary GC456000 Rev N</td>
<td>The applicability of the present STC is limited to Aircraft S/N 1100 to the limitations section in the TCCA-approved Airplane Flight Manual Supplement listed in associated technical documentation.</td>
<td>15/02/2010</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10023243</td>
<td>SOCATA</td>
<td>REV. 4</td>
<td></td>
<td>TB 21 TB 20</td>
<td>Installation of Garmin G500 EFIS interfaced with a dual Garmin GNS430/530(XM communication and navigation system, for IFR B-RNAV/P-RNAV operations with coupled or independent) Revision 3 of the STC including the extent to RNP LNAV and LPV approach operations. Revision 3 correcting typographic mistakes...</td>
<td>Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable: toolls certificate/approval: CS 23 The Certificate Basis for the original product remains applicable to this certificate/approval.</td>
<td>Installation instructions in &quot;STC 34.001.09.010, Edition 1 Revision 0, dated 05/10/2012. Instructions for continued airworthiness in &quot;STC 34.001.09.008, Edition 1, Revision 0, dated 05/05/2010. The software versions listed in STC 34.001.09.008, Edition 3, Revision 0 dated 12/07/2012 and Part...</td>
<td>23/10/2012</td>
<td>Active</td>
<td>10023243</td>
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<tr>
<td>10029250</td>
<td>AERO TECHNICS DESIGN LIMITED</td>
<td></td>
<td>CBAA UK RESPONSIBILITY</td>
<td>B767, L1000, 111, B777 B747-400, DC-8, DC-10, MD-11, A300, A310</td>
<td>Introduction of Intestate Airlines Airable Horse/Freight Container PN 73-90-343...</td>
<td>Various as applicable to aircraft types listed for installation. The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Certification Plan ATT0311 10029263 10029265</td>
<td>26/03/2010</td>
<td>Active</td>
<td>10029250</td>
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<tr>
<td>10029286</td>
<td>Airbus Engineering GmbH</td>
<td>REV. 2</td>
<td>SEE APPROVED MODEL LIST (AML)</td>
<td></td>
<td>Enabling of BFT functionalities on Garmin G9500 System (Revision of EASA STC 10025065)</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CS 23 Amendment 2 (not required by 21A.101); The certificated noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement: PO-0328-340000, Rev. 3 or later revisions of the above listed documents approved by EASA Description of Modifications: DE-0328-340000, Rev. 4. Engineering Work Order: EW-0328-340000, Rev. 3</td>
<td>none</td>
<td>10/02/2021</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<tr>
<td>BOMBARDIER INC.</td>
<td>10029275</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9323.</td>
<td>EASA FMA.008. The Certification Basis for the original product remains applicable to the certificate/approval. The certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>Transport Canada STC No. SA04-113 Issue 1 dated 28 October 2004.Aero Consulting Services Configuration Definition List D4A000 Revision/0 dated 03 March 2010.Instructions for Continued Airworthiness GC03290000-IFCA or later revisions of the above list...</td>
<td>17/03/2010</td>
<td>Active</td>
<td>10029275</td>
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<tr>
<td>BOMBARDIER AEROSPACE</td>
<td>10029282</td>
<td>Installation, Un-Upholstered Headliner, Rockwell Collins HJ5D.</td>
<td>EASA FMA.023. The Certification Basis for the original product remains applicable to the certificate/approval.</td>
<td></td>
<td>Transport Canada STC No. SA10-21 Issue 1 dated 04 March 2010.ACS-NAI Ltd. Modification Data Summary Number GS2300 Revision NC or later approved revision.For Instructions for Continued Airworthiness, refer to ACS-NAI Ltd.GS23070-ICA Revision NC or later approved revision or later revisions of the above...</td>
<td>17/03/2010</td>
<td>Active</td>
<td>10029282</td>
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<tr>
<td>FORCED AEROMOTIVE TECHNOLOGIES</td>
<td>10029290</td>
<td>Installation of BAR Air, LLC modified Teledyne Continental Motors550-N w/ MGL SR22-1-A, Rev. E or later FAA approved revisions. Installation requires EASA STC 10028703 to be installed.</td>
<td>CS-23.B. The Certification Basis for the original product remains applicable to the certificate/approval.</td>
<td></td>
<td>MDL-SR22-1-A, Rev. E Aircraft Flight Manual Supplement 22006 Rev. EIC A2206-30, Rev Eor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 200404CF (or subsequent revisions thereof)</td>
<td>17/03/2010</td>
<td>Active</td>
<td>10029290</td>
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<tr>
<td>FIELD AVIATION COMPANY INC.</td>
<td>10029315</td>
<td>Installation of a HF Antenna into the Vertical Stabilizer Leading Edge of an Allied Signal KHF-950 HF Transceiver.TCCA STC SA02-86, Issue 2</td>
<td>FAR-25 as per EASA FMA.183. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval.</td>
<td></td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/03/2010</td>
<td>Active</td>
<td>10029315</td>
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<tr>
<td>BELL TEXTRON PRAGUE, a s</td>
<td>10029324</td>
<td>Installation of Aspen EDF-1000 &quot;PRO&quot; and EDF-1000 &quot;PILOT&quot; GNS 430W GTX330, KR 87, KN 62A and ELT ME406 systems.</td>
<td>CS-23, Amdt. 1, dated 12 February 2009 including JAA Interim Poliservice INT/POL23/1 (HRF) and INT/POL23/3 (IE2).</td>
<td></td>
<td>L 200 D Flight Manual Supplements No.: FM 34 26/01 001 Primary Flight System EF01000 Pr dated Nov 27/00, FM 34 26/01 002 Primary Flight System EF01000 Pig. dated Nov 27/00, FM 34 56/01 001 GPS/WAAS NAV/COM/GPS GNS 430W, dated Nov 27/00, FM 34 56/01 002 Mode-S Transponder GTX 330, dated Nov 27/00.</td>
<td>18/03/2010</td>
<td>Active</td>
<td>10029324</td>
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### List of EASA Supplemental Type Certificates

#### EASA STC No. | STC Holder | Rev. level | Foreign/NAA STC No. | Model(s) | Description | Certification basis | Associated Technical Documentation | Limitations/Conditions | Date of Issue | Status | System Record
---|---|---|---|---|---|---|---|---|---|---|---

#### EASA STC No. | STC Holder | Rev. level | Foreign/NAA STC No. | Model(s) | Description | Certification basis | Associated Technical Documentation | Limitations/Conditions | Date of Issue | Status | System Record
---|---|---|---|---|---|---|---|---|---|---|---
10029332 | WESTERN AVIONICS | REV. 1 | TCCA STC SA03-53 ISSUE 1 | DHC-8-311, -314, -315, DHC-8-201, -202, -301, DHC-8-102, -103, -106, Primus 800 Weather Radar Installation | FAR-25 as per EASA.I.A. 191 The Certification Basis for the original product and the following or alternative airworthiness requirements are applicable to the certified equipment: | CS-23.B The Certification Basis for the original product remains applicable to this certificate/ approval Special Condition H-01 "Enhanced Airworthiness Programme for Aeronautic Systems - ICA on EWIS" as detailed in CRI. | As previously approved Weather Radar System installation is appreproachable for the installation. | 30/11/2010 | Active | 10029332

#### EASA STC No. | STC Holder | Rev. level | Foreign/NAA STC No. | Model(s) | Description | Certification basis | Associated Technical Documentation | Limitations/Conditions | Date of Issue | Status | System Record
---|---|---|---|---|---|---|---|---|---|---|---
10029335 | ACC COLUMBIA Jet Service GmbH | REV. 1 | T310P, T310Q, T310R | Alternative Supplemental Inspection Document (A-SID)Rev. 1 Transfer to ACC Columbia Jet Service GmbH | CS-23.B The Certification Basis for the original product remains applicable to this certificate/ approval | According to Gomolzig MDL Document No. MDCL21530, Revision 1512 2009 it is operated and maintained in accordance with approved Gomolzig Document No. 3A10.A-SID. Maintenance Manual Supplement 21530, Original Issue, December 2009, or later revisions of the above listed documents approved by EASA | As referred in limitations section of the 3A10_A-SID. Maintenance Manual Supplement 21530. | 20/01/2022 | Active | 10029335

#### EASA STC No. | STC Holder | Rev. level | Foreign/NAA STC No. | Model(s) | Description | Certification basis | Associated Technical Documentation | Limitations/Conditions | Date of Issue | Status | System Record
---|---|---|---|---|---|---|---|---|---|---|---
10029339 | AERO MEDICAL PRODUCTS Co. Inc. | FAA STC ST01188CH | CESSNA 501, 551 | Installation of Air Ambulance Conversion Kit | CS-23.B The Certification Basis for the original product remains applicable to this certificate/ approval | Installation Drawing List Report 006-1 FAA approved revisions | As referred in limitations section of the AFM/Prior to flight flight of each modified aircraft, and after each change to any of the communication, navigational, or Med Pack equipment, a ground EMI test must be conducted (EA Form 173 A copy of EA Form 173 must be kept with the aircraft records). | 19/03/2010 | Active | 10029339

#### EASA STC No. | STC Holder | Rev. level | Foreign/NAA STC No. | Model(s) | Description | Certification basis | Associated Technical Documentation | Limitations/Conditions | Date of Issue | Status | System Record
---|---|---|---|---|---|---|---|---|---|---|---
10029348 | APICAL INDUSTRIES, Inc. | FAA STC SR02271LA | 206L-3, 206L-4 | Bell 206 Cable Cutter Kit | FAR 27 The Certification Basis for the original product remains applicable to this certificate/ approval | -MD0206-21, revision A dated 10 November 2009 - Master Document List - ICAO26-21 revision N/C dated 16 July 2009 - Instructions for Continued Airworthiness or later revisions of the above listed documents approved by EASA | None | 19/03/2010 | Active | 10029348

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**Legend:**
- **EASA STC No.** | **STC Holder** | **Rev. level** | **Foreign/NAA STC No.** | **Model(s)** | **Description** | **Certification basis** | **Associated Technical Documentation** | **Limitations/Conditions** | **Date of Issue** | **Status** | **System Record**

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**Notes:**
- The list includes supplemental type certificates (STCs) for various models and installations.
- Certificate/ approval and remain applicable to this original product are unchanged and/ or emissions levels of the approvalThe certificated noise applicable to this certificate/ approval and remain applicable to this certificate/ approval.
- The Certification Basis for the original product remains applicable to this certificate/ approval.
- Installation must be in accordance with Western Avionics Engineering Order 90404-401, Rev. 2. Maintenance must be in accordance with Western Avionics Maintenance Manual Supplement 90404-701, Rev. 2 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/...
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<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>1002398</td>
<td>U.S. TECHNICAL CONSULTANTS, INC.</td>
<td></td>
<td></td>
<td>FAA STC ST02161A</td>
<td>Installation of navAero T.BagC2 Electronic Flight Bag</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>-FAA STC ST02161A- Master Drawing List 4913-25-00 rev B dated 05/02/2023 dated 05/11/08 later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>No Limitations</td>
<td>23/02/2010</td>
<td>Active</td>
<td>1002398</td>
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<tr>
<td>1002406</td>
<td>GAMA ENGINEERING Ltd.</td>
<td>REV. 1</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>Amendment to existing STC 1002406 to add TAWS-B functionality and include optional GAD34 autopilot interface unit.</td>
<td>C932 The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Master Data List MDL1112-02 Issue 1Change Statement CS1112-02 Issue 1Instructions for Continued Airworthiness IC1112-02 Issue 1 Flight Manual Supplement AFMS1112-02 Issue 1or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>03/01/2011</td>
<td>Active</td>
<td>1002406</td>
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<tr>
<td>1002416</td>
<td>STC TWENTY ONE LIMITED</td>
<td></td>
<td></td>
<td>BD-700-1A10</td>
<td>Customized Aircraft Completion - ASN S286</td>
<td>EASA IM.A.009 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CRI F-01 *Installation of Wireless System CRI D-11 &quot;Installation of Interior Door&quot; CRI D-12 <em>Isolated Compartments</em> The certifi...</td>
<td>STC Twenty One Ltd Master Document List No. S21.MDL-0665.STC Twenty One Ltd Airplane Flight Manual Supplement No. S21.FMS-0054 Instructions for Continued Airworthiness Midcoast Aviation Inc.Supplemental Maintenance Manual No. 35-02/06/14 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>24/03/2010</td>
<td>Active</td>
<td>1002416</td>
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<tr>
<td>1002418</td>
<td>BINDER MOTORENBAU GmbH</td>
<td>REV. 2</td>
<td></td>
<td>NIMBUS-3DM</td>
<td>Replacement of engine-control unit ILEC new Engine controller ILEC MCDU/NZDM can be installed in NIMBUS-3DM-Motorenbau with SOLO 26252 engines. Therefore, an adapted wiring will be introduced.</td>
<td>The Type and OfSD Certification Bases (CB) for the original product remains applicable to this certificate/approval The requirements for environmental protection and the associated certificate noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Technische Mitteilung (TM Triebenwerk 84) Solo-Umbau – Triebwerksbaumaßnahmen ILEC MCDU-NZDM und Motonelektrik, Ausgabe 10.02.2023 Zusatz zum Flughandbuch für den Motorenbau Nimbus SDM – Umbau von Rotax335C auf Solo 26252, Ausgabe 2 vom 10.02.2023 Zusatz zum Flughandbuch für den Motorenbau Nimbus...</td>
<td>None</td>
<td>05/04/2023</td>
<td>Active</td>
<td>1002418</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10029434</td>
<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>FAA STC DAS285SW-D</td>
<td>FALCON 900EX</td>
<td>EASA Validation of original FAA STC DAS285SW-D Revision 1. Installation of the Security System (Structural Provisions) in accordance with Drawing List Report No. FBEX-01201, Revision B dated 13th June 1997 or later revisions approved by EASA in accordance with FAA STC 200404/CF (or subsequent revisions). The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise/ or emissions levels of the original product unchanged and remain applicable to this certification/approval.</td>
<td>None.</td>
<td></td>
<td>24/02/2010</td>
<td>Active</td>
<td>10029434</td>
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<tr>
<td>10029450</td>
<td>TATENHILL AVIATION Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>M20, M20A, M20B, M20C, M20D, M20L, M20R, M20S, M20TN, M20E, M20F, M20G, M20J, M20K</td>
<td>Installation of Aspen EFD1000 Primary Flight Display (PFD)</td>
<td>The Certification Basis of the original product and the following: additional / alternative airworthiness requirements are applicable to this change. The certification noise emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>EASA approved Design Report with Installation Instructions, Doc. No. TAD.02209, Revision B, dated 25-Mar-2010, or later approved revisions.</td>
<td>FAA STC 24/03/2010</td>
<td>26/03/2010</td>
<td>Active</td>
<td>10029450</td>
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<td>10029451</td>
<td>BLR AEROSPACE, L.L.C.</td>
<td>FAA STC SA2054SE</td>
<td>C90A, C90GT, C90GTI</td>
<td>Installation of wheelset and associated changes in wing endplate structures and related system changes on Hawker Beechcraft Model C90A-C90GT and C90GTI - Revision 2 updates the STC holder's address.</td>
<td>The Certification Basis of the original product remains applicable to this certificate/approval.</td>
<td>Installation in accordance with BLR Master Drawing List, Document/Reference BLR KA90W-002, rev. J, dated March 02, 2010, or later approved revisions.</td>
<td>FAA STC 22/05/2010</td>
<td>Active</td>
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<td>10029476</td>
<td>ECT INDUSTRIES</td>
<td>AS356F2, AS356N, AS356NP, AS356E, AS356F, AS356F1</td>
<td>STC DTS AS 356 - installation of a Tracking system.</td>
<td>STC DTS AS 356 - installation of a Tracking system.</td>
<td>The Certification Basis for the original product and the following: additional / alternative airworthiness requirements are applicable to this certificate/ approval. The certified noise emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>FAA/AFMS-DTS-356-EN, rev. 0 - Rotorkraft Flight Manual Supplement or later approved revisions of the above listed products approved by FAA STC DAS285SW-D Revision 1 as amended by the FAA on 14th June 1997 or later revisions of the above listed documents approved by EASA in accordance with FAA STC DAS285SW-D Revision 1.</td>
<td>None.</td>
<td>26/03/2010</td>
<td>Active</td>
<td>10029476</td>
<td></td>
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<td>10029480</td>
<td>SOCIETE D'ETUDE ET DE GESTIONAERON AUTIQUE (SEIGA)</td>
<td>AS350 B2</td>
<td>Installation system for &quot;EC120 type&quot; seats on Eurocopter AS350B2 modification H-06-09-Siege</td>
<td>Installation system for &quot;EC120 type&quot; seats on Eurocopter AS350B2 modification H-06-09-Siege.</td>
<td>As per EASA TCDS R.008 Issue 03, 12 October 2007. The Certification Basis for the original product remains applicable to the certificate/approval. The certified noise emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Installation system for EC120 seats on Eurocopter AS350B2 modification H-06-09-Siege - GS. 0 dated February 25th 2010.</td>
<td>None.</td>
<td>26/03/2010</td>
<td>Active</td>
<td>10029480</td>
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<td>STC Holder</td>
<td>Active System</td>
<td>System Active</td>
<td>STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>ICELANDAIR TECHNICAL SERVICES</td>
<td></td>
<td></td>
<td>STC9481</td>
<td>B757-200</td>
<td>Navigation - Dependent Position Determining - Activation of the ADS-B Out function of the Transponder system.</td>
<td>Certification Basis for the original product and the following/alternative or equivalent airworthiness requirements are applicable to the STC Certificate/approval. Applicability is limited to comply with CS25.17.</td>
<td>FAA-AP-757-3450-30 Mirrored: Continued Airworthiness Instructions CAI-757-3450-30 rev. 0 Continued Airworthiness Instructions CAI-757-3450-30 rev. 0 Change Approval ORA-757-3450-30 rev. 0 Certification Plan 757-34-50-30 rev. 0Drawing DWG-757-3450-30 rev. 0 Drawing DWG-757-3450-30 rev. 0</td>
<td>None.</td>
<td>29/03/2010</td>
<td>Active</td>
<td>10029481</td>
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<tr>
<td>GAMA ENGINEERING LIST</td>
<td></td>
<td></td>
<td>STC9459</td>
<td>304B</td>
<td>Navigation/communications systems upgrade.</td>
<td>Certification Basis for the original product and the following/alternative or equivalent airworthiness requirements are applicable to the STC Certificate/approval.</td>
<td>AMCS20-4APDOAP 1ICA-1130-01 Issue 1FA-1130-01 Issue 1MDO-1130-01 Issue 1AEMS-1130-01 Issue 1AMS-1130-01 Issue 2AMS-1130-01 Issue 2AMS-1130-01 Issue 2AMS-1130-01 Issue 2AMS-1130-01 Issue 2AMS-1130-01 Issue 2AMS-1130-01 Issue</td>
<td>None.</td>
<td>29/03/2010</td>
<td>Active</td>
<td>10029499</td>
</tr>
<tr>
<td>M&amp;D FLUGZEUGBAU GmbH &amp; CO. KG</td>
<td>(EASA.A.S.02064)</td>
<td></td>
<td>STC9513</td>
<td>304B</td>
<td>Modification of Glasfluegel 304B, SN 324 - Retrofitting with winglets - Modified wing tips: Winglets.</td>
<td>Certification Basis for the original product and the following/alternative or equivalent airworthiness requirements are applicable to the STC Certificate/approval.</td>
<td>STC &quot;Nachrüsten mit Winglets/Retrofitting with Winglets&quot; of sailplane Glasfluegel 304B, S/N 324 by APDOAP</td>
<td>None.</td>
<td>29/03/2010</td>
<td>Active</td>
<td>10029513</td>
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<tr>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td></td>
<td>STC9540</td>
<td>50 (MF50)</td>
<td>Runway Awareness and Advisory System (RAAS). Smart Runway Activation i.e. Jet Aviation Master Data List BF5E3461-001 Revision B or later/EASA Approved revisions.</td>
<td>Certification Basis for the original product and the following/alternative or equivalent airworthiness requirements are applicable to the STC Certificate/approval. In addition compliance has been shown i.e. EASA CRI F-01 Issue 02 &quot;Activation of the Runway Awareness and Advisory System (RAAS)&quot; that i.e.</td>
<td>See: &quot;Jet Aviation Master Data List BF5E3461-001 Revision B and Jet Aviation Aeroplane Flight Manual Supplement BF5E3461-201-01 Revisions on IR Jet Aviation Engineering Order (installation) BF5E3461-344600Revisions on Bor later revisions of the above listed documents approved by EASA</td>
<td>None.</td>
<td>30/03/2010</td>
<td>Active</td>
<td>10029540</td>
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<tr>
<td>AVIATION TECHNICAL SERVICES, Inc</td>
<td>(EASA.IMA.S.02553)</td>
<td></td>
<td>STC9541</td>
<td>737-400</td>
<td>Installation of a Collins Traffic Alert and Collision Avoidance System (TCAS II) Transmitter/Receiver per Goodrich ATS Inc Top Drawing 90046Revision F dated 05 January 2001.</td>
<td>Certification Basis for the original product remains applicable to the STC Certificate/approval. The certificated noise (and/or emissions levels of the original product) have remained unaltered and remain applicable to this certificate/approval.</td>
<td>- Goodrich ATS Inc Top Drawing 90046 Revision dated 05 January 2001 - Airplane Flight Manual Supplement Document D2000-181 Original Revision Approved by FAA on 22 February 2001 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/84/EC (or subseq.)</td>
<td>None.</td>
<td>30/03/2010</td>
<td>Active</td>
<td>10029541</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Rev.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>SASTED AVIONICS, Inc.</td>
<td>FAA STC ST01950LA</td>
<td>S550</td>
<td>Installation of Sanded Avionics S5450 Primary Attitude Display and Sanded Avionics S5450 Primary Navigation Display.</td>
<td>The Certification Basis for the original product remains applicable, and the associated technical documents references.</td>
<td>None.</td>
<td>31/03/2010</td>
<td>Active</td>
<td>10029542</td>
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<td>CASCADE AEROSPACE Inc.</td>
<td>TCCA STC SA01-40</td>
<td>DHC-8-3.1, -302 DHC-8-201, -202 DHC-8-102, -103, -106</td>
<td>Ground Air Connection ModificationTCCA STC SA01-40, as is, dated 14 March 2001</td>
<td>FAR 25 as per EASA.M.A. 191</td>
<td>Installation of Air Ground Connection Modification is to be carried out in accordance with Conair Aerospace Engineering Order EO 2846, Rev. 1 dated 01 August 2002. Maintenance is to be carried out in accordance with Conair Aerospace Engineering Maintenance Manual Supplement MMS 2864-1, Rev. NIL, 1st...</td>
<td>31/03/2010</td>
<td>Active</td>
<td>10029545</td>
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<tr>
<td>365 AEROSPACE DOS LIMITED</td>
<td>REV. 3</td>
<td>737-600-700-800/900</td>
<td>Introduction of a Dual Battery SystemRev. 3. Transfer of STC from 365 Aerospace Ltd. to 365 Aerospace DDS Ltd.</td>
<td>The Certification Basis for the original product remains applicable, and the associated technical documents references.</td>
<td>Aerospace Design ((\text{&amp;}1)) Engineering Consultants Modification SummaryMOD-0702 Revision C dated 22nd March 2010.Aerospace Design ((\text{&amp;}1)) Engineering Consultants EZA-0702 Revision Aplled 15th November 2011</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change will not introduce onboard upset into the airworthiness of the product.</td>
<td>05/08/2019</td>
<td>Active</td>
<td>10029548</td>
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<tr>
<td>MECAER AVIATION GROUP S.p.A.</td>
<td>MDS HELICOPTERS 369HS</td>
<td></td>
<td>Installation of Emergency Locator Transmitter (ELT), Artex C406-2HM, P/ N369-25543-101</td>
<td>The Certification Basis for the original product remains applicable, and the associated technical documents references.</td>
<td>-SEI RFNS - 369HS-1-1 at Rev A or later EASA approved revision Report - SEI Description and CCL - ETL C406-2 Rev B dated 24 March 2010or later EASA approved revision Report - SEI Notice of Design Change - MDC 369-0350on later revisions of the above listed documents approved by EASA</td>
<td>Not Applicable</td>
<td>31/03/2010</td>
<td>Active</td>
<td>10029549</td>
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<td>STC Holder</td>
<td>Active Status</td>
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<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>ST ENGINEERING AEROSPACE LTD.</td>
<td>Active</td>
<td></td>
<td>A320</td>
<td>Passenger Compartment Reconfiguration</td>
<td>In accordance with the EASA TCDS A.064 plusnThe Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/approval-CRI H-01/ Special Condition H-01. <em>Enhanced Airworthiness Programme forAeroplane Systems - ICAT...</em></td>
<td>A320-01PA-S TCST Certification Plan, ref. doc. No. 38404031/AED/002.Revision C, dated 31st March 2010</td>
<td>None</td>
<td>31/03/2010</td>
<td>Active</td>
<td>10029550</td>
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<tr>
<td>JET AVIATION AG, BASEL</td>
<td>REV. 1</td>
<td></td>
<td>B737-800</td>
<td>EMS SATCOM AMT-3800 Installation</td>
<td>The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/approval-CRI H-01/ Special Condition H-01. <em>Enhanced Airworthiness Programme forAeroplane Systems - ICAT...</em></td>
<td>Jet Aviation Master Data List No. BB737J3263-001, Rev. C, 07 06 2010 or later revisions of the above listed documents approved by EASA</td>
<td>Not Applicable</td>
<td>12/07/2010</td>
<td>Active</td>
<td>10029553</td>
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<tr>
<td>JET AVIATION AG, BASEL</td>
<td>REV. 1</td>
<td></td>
<td>AIRBUS A319-115</td>
<td>Airbus A319-115 “BCJ03513” VIP Cabin Interior CompletionRevision 1: Update of the instructions for Continued Airworthiness</td>
<td>The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable tothis certificate/approval-CRI H-01/ Special Condition H-01. <em>Enhanced Airworthiness Programme forAeroplane Systems - ICAT...</em></td>
<td>Airplane Flight Manual Supplement BCJ03513-01-01 Revision IR from 15March 2010Instructions for Continued AirworthinessBCJ03513-15 0-01 Revision C from 02 Aug. 2010Master Data List BCJ03513-01 Revision C from 03 Aug. 2010Declaration of Compliance CAF 3607 from 03 Aug. 2010 or later revisions of the...</td>
<td>Not Applicable</td>
<td>04/08/2010</td>
<td>Active</td>
<td>10029554</td>
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<tr>
<td>AIR COMM CORPORATION LLC</td>
<td>FAA STC</td>
<td></td>
<td>EC 130 B4</td>
<td>Installation of Cabin Air Conditioning System</td>
<td>The Certification Basis for the original product remains applicable tothis certificate/approval The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval</td>
<td>- Air Comm Corporation Master Drawing List Report No DL-EC-130, Revision U, dated 12/19/2009- Report EC130-03-10, dated March 16, 2015- Rotorcraft Flight Manual Supplement Document No EC130B4-1 dated April, 2004- Rotorcraft Flight Manual Supplement Document No EC130B4-2 dated March05, 2006 (Condensed...</td>
<td>None</td>
<td>12/04/2010</td>
<td>Active</td>
<td>10029555</td>
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<td>TRUATLANTIC MFG., LLC</td>
<td>FAA STC</td>
<td>REV. 2</td>
<td>U206, U206A, U206B, U206C</td>
<td>Installation of Remote Oil Cooler in accordance with Atlantic Aero, Inc. Master Drawing List 16693001 Rev. Basic dated December 5, 1997, or later FAA Approved revision</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Approved Supplemental Flight Manual dated April 15, 1998 for the following Models and Serial Numbers Caravan U206E through U206G, TU206E through TU206G (Serial Numbers U206144S through U206144H) Document 16693100FF FAA Approved Pilots Operating Handbook and FAA Approved Flight ManualSupplement date...</td>
<td>a) This approval should not be extended to other aircraft of this modelon which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse effect...</td>
<td>28/04/2016</td>
<td>Active</td>
<td>10029557</td>
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<td>STC Holder</td>
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<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>LEES AVIONICS Ltd.</td>
<td></td>
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<td>COMMANDER 112, 114</td>
<td>Installation of Garmin G500/G600 PFD/ND</td>
<td>-Master Data List MDL1097-01 Issue 3- Change Statement CS1097-01 Issue 2- Instructions for Continued Airworthiness ICA1097-01 Issue 3-Flight Manual Supplement AFMS1097-01 Issue 3or later revisions of the above listed documents approved by EASA.</td>
<td>This change is limited to the configuration defined in Change StatementCS1097-01 Issue 2 or later approved revision.</td>
<td>01/04/2010</td>
<td>Active</td>
<td>10029565</td>
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<td>AIR METHODS CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SR00075DE-D</td>
<td>S-76A, S-76B, S-76C</td>
<td>Installation of Articulated Patient Loading System (APLS) Note: The STC amendment adds an ALLFA litter as an option.</td>
<td>As described in FAA TCDS HTN/EThe Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to thes certificates/approvalCS-23EASA GM/G/001 has been applied. The certificated noise and/or emissions levels of the original product were unchanged and remain applicable.</td>
<td>Required pre-modification Sikorsky cabin medical interior CAA ref. no.03075/10E/001-1</td>
<td>19/08/2011</td>
<td>Active</td>
<td>10029569</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>Husky A-1 Silencer STC Ext. To Variant A, A-1C</td>
<td>CS-23 CTThe Certification Basis for the original product remains applicable to thes certificates/approval.</td>
<td>-Rotorcraft Flight Manual Supplement No. 080-0476, revision4, datedJune 9, 2011or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/03/CF (or subsequent revisions of this decision)-Master Drawing List S-76 APLS w/ Pedestal and AMC litter, Report N...</td>
<td>As referred in Limitations section of the AFM-S:Airplane Flight ManualSupplementor later approved revision.</td>
<td>20/01/2022</td>
<td>Active</td>
<td>10029570</td>
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<td>AIR ALLIANCE GmbH</td>
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<td>LEARJET 35A</td>
<td>Installation of new avionics (FDR, TAWS, FMS, RDR,VDI)</td>
<td>As referred in Limitations section of the AFM-S:Airplane Flight Manual Supplement</td>
<td>Approved valid for s/n 35A-616 only</td>
<td>02/04/2010</td>
<td>Active</td>
<td>10029583</td>
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<td>BOMBARDIER INC.</td>
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<td>TCAC STC SA10-19</td>
<td>Installation, Upgrade from Swift 63 to Swift Broadband</td>
<td>EASA 0610L The Certification Basis for the original product remains applicable to thes certificates/approval The certificated noise and/or emissions levels of the original product were unchanged and remain applicable to this certificate/approval.</td>
<td>-TCOA STC No. SA10-19 Issue 1 dated 06 February 2010-TCA SoC ref. C-10-0232 dated 24 March 2010-ACS-NAI Ltd. Modification Data Summary Number GS41000 Revision NC ohlter approved revision-Instructions for Continued Airworthiness GS41071 Revision NC or lateremission or later revisions of the above...</td>
<td>None</td>
<td>02/04/2010</td>
<td>Active</td>
<td>10029589</td>
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<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>R65 PRODUCTS, INC.</td>
<td>REV 1</td>
<td>US SR09178RC</td>
<td>600N</td>
<td>Integrated Flight Systems Vapor Cycle Air Conditioning System</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>None</td>
<td></td>
<td>29/06/2010</td>
<td>Active</td>
<td>10029611</td>
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<td>MCCAULEY PROPELLE Systems</td>
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<td>FAA STC SA0151WI</td>
<td>REV. 1</td>
<td>FAA STC ST0154AW</td>
<td>750</td>
<td>Installation of window, wing structural reinforcement, and related system changes.</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval, except where amended by additional later amendments if indicated on FAA STC. There requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certification/approval.</td>
<td>Winglet Technology Master Drawing List: D10-2000, Rev. B, dated 16 June 2009, or later FAA approved revision. FAA approved/Airplane Flight Manual (AFM) Supplements and Weight and Balance Manual Supplements.</td>
<td>Prior to installation of this design change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce adverse effects upon the airworthiness of the product. See Winglet Technology Master Supplement.</td>
<td>25/10/2018</td>
<td>Active</td>
<td>10029626</td>
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<tr>
<td>BOMBARDIER INC.</td>
<td>REV. 1</td>
<td>FAA STC SA014-11</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9336</td>
<td>EASA M.JA.009: The Certification Basis for the original product remains applicable to this certification/approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>- Transport Canada SoC reference C-10-0075 dated 07 April 2010; - TCCA STC No. SA04-113 Issue 1 dated 28 October 2004; - Aeroc Consulting Services Configuration Definition List D954000 Revision ED dated 31 March 2010; - Instructions for Continued Airworthiness C933-4000-000-10007 (EASA) 5/28/2003; - FAA approved revision FAA approved/Airplane Flight Manual (AFM) Supplements and Weight and Balance Manual Supplements.</td>
<td>Prior to installation of this design change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce adverse effects upon the airworthiness of the product. See Winglet Technology Master Supplement.</td>
<td>08/04/2010</td>
<td>Active</td>
<td>10029628</td>
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<td>BERINGER AERO</td>
<td>REV. 7</td>
<td>PC-6B2-H2, PC-6A2-H2, PC-6B1-H2</td>
<td></td>
<td>Installation of Beringer brakes Revision 1: Update with an optional main wheel. Revision 2: Update with an optional axle. Update of manuals. Revision 3: Update with an additional brake disc. Revision 4: Update with an additional brake pad. Revision 5: Update with wear limit modification. Revision 6:</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>Modification file: DM-SC-002 rev 0 dated 06 Dec. 2022: Installation and Instructions for Continued Airworthiness ICA-STC-002 rev 0 dated 31 Dec. 2022; or later revisions of the above list documents approved by EASA.</td>
<td>Prior to installation of this design change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce adverse effects upon the airworthiness of the product.</td>
<td>21/12/2022</td>
<td>Active</td>
<td>10029629</td>
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<td>KEYSTONE HELICOPTER CORPORATION</td>
<td>REV. 7</td>
<td>FAA STC SR0258NY-D</td>
<td>S-76C</td>
<td>Installation of Goodrich Auxiliary Utility Hose</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certification/approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>-RFA5 FMA1246 Rev 6 dated 25 Oct 2005; or later revisions of the above list documents approved by EASA.</td>
<td>As specified in the approved RFA5</td>
<td>08/04/2010</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>WIPARE Inc.</td>
<td>FAA STC</td>
<td>SESNA 206 SERIES</td>
<td>B737-300</td>
<td>Installation of Wipaire, Inc. wing tip extensions in accordance to FAA STC SESNA 206 SERIES</td>
<td>The Certification Basis for the original product remains applicable to the certificate's approval. The requirements for environmental protection and the associated certificate, noise and or emissions levels of the original product are unchanged and remain applicable to this certificate's approval.</td>
<td>Wipaire Inc. (Master) Drawing List number W-MD26, dated April 30, 2001, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/OF (or subsequent revisions of the decision)</td>
<td>08/04/2010</td>
<td>Active</td>
<td>10229642</td>
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<td>DECA AVIATION ENGINEERING S.A.</td>
<td>TCCA STC</td>
<td>O-LSA09-371/D</td>
<td>B737-800</td>
<td>Interior Reconfiguration &amp; miscellaneous Cabin Alterations / Modification and relocation of the passenger seats along with relocation/installation of emergency equipments.</td>
<td>The Certification Basis for the original product remains applicable to the certificate's approval. The certificate, noise and or emissions levels of the original product are unchanged and remain applicable to this certificate's approval.</td>
<td>- TCCA STC Number: O-LSA09-371/D Issue 1 dated Feb 03, 2009. DECA Modification summary MS0165 dated Feb 03, 2009, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/OF (or subsequent revisions of the decision)</td>
<td>Valid for MSN 30292 only</td>
<td>09/04/2010</td>
<td>Active</td>
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<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 2</td>
<td>737-300</td>
<td>B737-300</td>
<td>Transfer of STC from the original aircraft to a different aircraft type.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate. The requirements for environmental protection and the associated certificate, noise and or emissions levels of the product are unchanged and remain applicable to this certificate's approval.</td>
<td>- Modification Summary Sheet: MOD-11348-0001 Issue 3 dated 16 July 2015. Master Document List: MDL-11348-0001 Issue 3 dated 15 July 2015. Aircraft Flight Manual Supplement: 11348-AMF-0001 Issue 2. Enhanced Zonal Analysis: EZA-11348-0001 Issue 3 dated 15 August 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>Limited to following aircraft S/N only: 737-300 MSN 23774. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effects upon the airworthiness of the product.</td>
<td>10/11/2021</td>
<td>Active</td>
<td>10229662</td>
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<tr>
<td>DASSAULT FALCON SERVICE</td>
<td>FALCON 900EX</td>
<td>F900EX S/N 111</td>
<td>FALCON 900EX</td>
<td>Certification Basis for the original product remains applicable to the certificate's approval. Compliance has been shown in accordance with the requirements laid down by EASA in CNI F-002 “In-Seat Power Supply Systems (SPSS)” issue 2 and O-LSA09-371/D Issue 1 dated 03/07/2024</td>
<td>- DFS Change Approval Sheet DFS 9565-CAS Issue A dated 30th March 2010-DFS Master Document List DFS-DE-17- FE-9565, Issue A dated 16th October 2009-DFS Modification Sheet DFS 9565, Issue A dated 29th October 2009, or later revisions of the above listed documents approved by EASA</td>
<td>This EASA design change airworthiness approval is applicable on Falcon 900EX serial number aircraft 111</td>
<td>12/04/2010</td>
<td>Active</td>
<td>10229669</td>
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<td>COMPAGNIA GENERALE RIPRESEAREEES p.A. - COR S.p.A.</td>
<td>PA 31-350</td>
<td>GPS Garmin 430 for IFR</td>
<td>EASA CS 23 Amendment 1 The Certification Basis for the original product remains applicable to the certificate's approval.</td>
<td>APMS PIN CDR P003-00-SICA PIN CDR P003-00 for later revisions of the above listed documents approved by EASA</td>
<td>The installation of EASA approved modification EASA.C.07903 is required.</td>
<td>12/04/2010</td>
<td>Active</td>
<td>10229675</td>
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<td>Rev. level</td>
<td>Foreign/NA</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10029677</td>
<td>LUTFHANS TECHNIN AG</td>
<td>Active</td>
<td>Active</td>
<td>Active</td>
<td>Fuselage - Structural provisions and VHF antenna installation. This STC installs Satcom and VHF antennas as well as the provision for future support of a Satcom system installation.</td>
<td>The Certification Basis for the original product and the follow-on/additional/alternative airworthiness requirements are applicable to the applicable STC certification basis.</td>
<td>Limited to: MSN 484</td>
<td>12/04/2010</td>
<td>Active</td>
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<td>10029680</td>
<td>ARLIFT A/S</td>
<td>FAA STC</td>
<td>AS365N SERIES</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement, FMS-365-25-105 Revision 3 or later revisions of the above listed document approved by EASA - Instructions for Continued Airworthiness, ICA-365-25-105-01 Revision 2</td>
<td>None</td>
<td>12/04/2010</td>
<td>Active</td>
<td>10029680</td>
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<td>10029687</td>
<td>RUECKER AEROSPACE GmbH</td>
<td>FAA STC</td>
<td>B747-400</td>
<td>Installation of 2 single B/E Aerospace DS9000 steam ovens and 2 GCM0000 Oven Control Modules per B/E Aerospace Engineering OrderEO-253-10002-001 Rev. A, dated 12/04/2010</td>
<td>The Certification Basis for the original product and the follow-on/additional/alternative airworthiness requirements are applicable to the applicable STC certification basis.</td>
<td>Limited to: MSN 484</td>
<td>13/04/2010</td>
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<td>10029701</td>
<td>CESKA LETECKA SERVISNI A.S.</td>
<td>FAA STC</td>
<td>SUKHOI SU-31</td>
<td>Avionics upgrade.</td>
<td>The Certification Basis for the original product and the follow-on/additional/alternative airworthiness requirements are applicable to the applicable STC certification basis.</td>
<td>Limited to: MSN 484</td>
<td>14/04/2010</td>
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<td>CAA UK</td>
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<td>ERWIN AERO INTERNATIONAL LTD.</td>
<td>10029711</td>
<td>TC A CAA</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10029716</td>
<td>airbus engineering GmbH</td>
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<td></td>
<td>Enabling of SVT Functionalities on Garmin G5/600 Avionics System</td>
<td>Flight Manual Supplement PO-0491-340000 Rev. 2</td>
<td>Not Applicable</td>
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<td>GAMA ENGINEERING Ltd</td>
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<td>Installation of Garmin G5600 G600 EFIS PFD/ND.</td>
<td>Master Data List MDL1050-02 Issue 1</td>
<td>This change is limited to the configuration defined in Change Statement 1050-02 Issue 1 or later approved revision.</td>
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<td>GAMA ENGINEERING Ltd</td>
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<td>Installation of Silver Jodide Generators on Cessna 421B</td>
<td>- Flight Manual Supplement SPOH_16351 rev 1 15032 009 (dated 15.03.2009)</td>
<td>Not Applicable</td>
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<td>1002973</td>
<td>B/E AEROSPACE LIMITED</td>
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<td>Main battery No. 2 relocation/Validation of the TCCA STC SA10-4</td>
<td>- Modification Data Summary, Doc. No. G409600 Rev. C- PIlot’s Operating Handbook Supplement, Doc. No. G409691 Rev. N/IC/CA</td>
<td>None.</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<td>BOMBARDIER INC.</td>
<td>TC07-01A16</td>
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<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASIN 9341</td>
<td>EASA EMA.2-009 The Certification Basis for the original product remains applicable to this certificate approval</td>
<td></td>
<td>Restricted to ASIN 9341 and I. Refer to the Limitations sections of the required AFM Supplement and the referenced FC/OMS 2. WLAN system is compatible only to IEEE 802.11(b).</td>
<td>20/04/2010</td>
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<td>FAA STC REF SA02277LA</td>
<td>B300 KING AIR</td>
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<td>Installation of High Definition Audio Video (HDAV-3000) LRU/Kit as aninterface with the Rockwell Collins Venue Cabin Management System.</td>
<td>FAR 21 The Certification Basis for the original product remains applicable to this certificate approval</td>
<td></td>
<td>Only applicable to King Air Model B300 aircraft equipped with the Rockwell Collins Venue Cabin Management System.</td>
<td>20/04/2010</td>
<td>Active</td>
<td>10029794</td>
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<td>FAA STC SA02042SE</td>
<td>CESSNA 208</td>
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<td>Aircraft Payload Extender (APE) STOL System for Cessna 208 and 208A inacc. to FAA STC SA02042SE</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval</td>
<td></td>
<td>Note 1: This modification is also applicable to the following aircraft that have been modified with Wipaire, Inc. STC No. SA146CH (875 SHP PT6A-114A Engine Installation). Model, Serial Number, ..., Note 2088...20880061</td>
<td>06/08/2012</td>
<td>Revised</td>
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<td>FAA STC SA02042SE</td>
<td>CESSNA 208</td>
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<td>Aircraft Payload Extender (APE) STOL System for Cessna 208 and 208A inacc. to FAA STC SA02042SE</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval</td>
<td></td>
<td>Only applicable to Cessna Model 208 aircraft equipped with Rockwell Collins Venue Cabin Management System.</td>
<td>06/08/2012</td>
<td>Active</td>
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<tr>
<td>FAA STC SA02042SE</td>
<td>CESSNA 208</td>
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<td>Aircraft Payload Extender (APE) STOL System for Cessna 208 and 208A inacc. to FAA STC SA02042SE</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval</td>
<td></td>
<td>Only applicable to King Air Model B300 aircraft equipped with Wipaire, Inc. STC No. SA146CH (875 SHP PT6A-114A Engine Installation). Model, Serial Number, ..., Note 2088...20880061</td>
<td>06/08/2012</td>
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<td>FAA STC SA02042SE</td>
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<td>Aircraft Payload Extender (APE) STOL System for Cessna 208 and 208A inacc. to FAA STC SA02042SE</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval</td>
<td></td>
<td>Only applicable to King Air Model B300 aircraft equipped with Wipaire, Inc. STC No. SA146CH (875 SHP PT6A-114A Engine Installation). Model, Serial Number, ..., Note 2088...20880061</td>
<td>06/08/2012</td>
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<td>Limitations/Conditions</td>
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<td>AVIATION FABRICATORS, Inc.</td>
<td>10029817</td>
<td>REV. 1</td>
<td>FAA STC ST01572W</td>
<td>400A</td>
<td>400A</td>
<td>Change to STC 10029817 Beechjet 400 Cabin Configurations, Divan Installations.</td>
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<td>28/05/2013</td>
<td>Active</td>
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<td>VIH AEROSPACE Inc.</td>
<td>10029820</td>
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<td>TCCA STC SH06-26</td>
<td>S-92A</td>
<td>Auxiliary Fuel Tank Installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>V592-371-FMS revision 5 dated 22 August 2007 - Flight manual Supplement</td>
<td></td>
<td>22/04/2010</td>
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<td>AEROCOUSTICS AIRCRAFT SYSTEMS, Inc.</td>
<td>10029821</td>
<td>REV. 2</td>
<td>FAA SA018055E</td>
<td>208B</td>
<td>Fabrication and installation of Aerocoustics Aircraft Systems, Inc. (AASS) Aircraft Payload Extender STOL System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>V592-371-MDL revision 5 dated 17 September 2007 - Master Drawing List AF-475, Rev D, dated 22 August 2007 - Flight manual Supplement or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/022/CF (or subsequent revisions of this decision)</td>
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<td>09/04/2021</td>
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<td>AERO CONSULTING SERVICES LTD.</td>
<td>10029833</td>
<td></td>
<td>TCCA STC SA10-9</td>
<td>CL600-2B16, CL600-4A1, CL600-2A12,</td>
<td>Transport Canada STC No. SA10-9 Issue 1 dated 08 February 2010, ACS NAV Ltd. Modification Data Summary Number G373000 Revision M1, other approved revision instructions for Continued Airworthiness G373071, Revision A, and G373072 (Rev 2), or later approved revisions or later revisions of the above...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>EASA IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td></td>
<td>23/04/2010</td>
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<td>GVH AEROSPACE LIMITED</td>
<td>10029836</td>
<td></td>
<td>BOEING 737-800</td>
<td>164</td>
<td>Passenger Layout.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>- Compliance Document BAC7377131 Issue 7, Instructions for Continued Airworthiness ICA/114/881 Issue for later revisions of the above listed documents approved by EASA</td>
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<td>23/04/2010</td>
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<td>EASA STC No.</td>
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<td>Associated Technical Documentation</td>
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<td>10029849</td>
<td>AIRCRUISES COMPANY</td>
<td>FAA STC SR02538NY</td>
<td>BELL 407</td>
<td>Bell 407 Emergency Float Kit P/N 67432-101/0 Note: This STC is for the installation of the Pneumatic Activation System only.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>EASA STC no. 03/07/2024</td>
<td>FAA STC STC02783CH</td>
<td>Issued by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions or this decision)</td>
<td>26/04/2010</td>
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<td>10029850</td>
<td>AVIATION TECHNICAL SERVICES, Inc.</td>
<td>FAA STC0617E-5</td>
<td>737-800</td>
<td>Interior Reconfiguration - installation of an Interior Arrangement, Entertainment System Components and Avionics Components.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAA STC STC02784CH</td>
<td>FAA STC SA00961WI</td>
<td>Issued by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions or this decision)</td>
<td>26/04/2010</td>
<td>Active</td>
<td>10029850</td>
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<td>10029856</td>
<td>AVIONIK ENTWICKLUNGSGMBH</td>
<td>FAA STC0296CH</td>
<td>PA-206-053-184-122</td>
<td>Bell 407 Emergency Float Installation Kit is used in conjunction with the float bags approved equivalents. The ACC and 206-053-184-122 are required for operation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAA STC STC02787</td>
<td>FAA STC SA00961WI</td>
<td>Issued by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions or this decision)</td>
<td>26/04/2010</td>
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<td>10029869</td>
<td>AVIATION FABRICATORS, Inc.</td>
<td>FAA STC NO. SA00961WI</td>
<td>PA-42, PA-42-720, PA-42-1000, PA-317, PA-317T, PA-31 PA-319, PA-31, PA-31T, PA-31T1 PA-31-325, PA-31-350</td>
<td>Stryker Installation per Aviation Fabrickators, Inc. STC Drawing List, Doc. No. AF-204</td>
<td>Design Non-significant change: The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC STC02788CH</td>
<td>FAA STC SA00961WI</td>
<td>Issued by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions or this decision)</td>
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<td>10029872</td>
<td>ELLIOTT AVIATION</td>
<td>FAA STC STC02784CH</td>
<td>BEECHJET 400400A</td>
<td>Installation of Honeywell Digital Flight Data Recorder/Validation of FAA STC STC02784CH</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAA STC STC02787</td>
<td>FAA STC SA00961WI</td>
<td>Issued by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions or this decision)</td>
<td>26/04/2010</td>
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<td>10029873</td>
<td>ELLIOTT AVIATION</td>
<td>FAA STC STC02783CH</td>
<td>BEECHJET 400</td>
<td>Installation of an Artec C406-2 Emergency Locator Transmitter (ELT) Validation of FAA STC STC02783CH</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAA STC STC02787</td>
<td>FAA STC SA00961WI</td>
<td>Issued by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions or this decision)</td>
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<td>ARMSTRONG AEROSPACE Inc.</td>
<td>B737-300, B737-500</td>
<td>B737-300, B737-500</td>
<td>Activation of Honeywell Runway Awareness and Advisory System (RAAS)</td>
<td>None</td>
<td>FAA STC/STC No. B737-300, B737-500</td>
<td>28/04/2010</td>
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<td>AEROLITE AG</td>
<td>EC135 T2(CPD5), EC135 T2+</td>
<td>EC135 T2(CPD5), EC135 T2+</td>
<td>Installation of an Espresso Maker</td>
<td>None</td>
<td>FAA STC/STC No. EC135 T2(CPD5), EC135 T2+</td>
<td>Limited to: S/N 870 and subsequent</td>
<td>08/11/2012</td>
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<td>BEECHCRAFT CORPORATION</td>
<td>800XP, 850XP, 900XP</td>
<td>800XP, 850XP, 900XP</td>
<td>Installation of an Espresso Maker</td>
<td>None</td>
<td>FAA STC/STC No. 800XP, 850XP, 900XP</td>
<td>Limited to: S/N 870 and subsequent</td>
<td>03/05/2010</td>
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<td>10029922</td>
<td>STC Holder</td>
<td>REV. 1</td>
<td>TCCA LSTG</td>
<td>B737-800</td>
<td>Sliding Carpet Loading System Removal/Deactivation</td>
<td>Revision 1: Removal of limitation on aircraft MSN.</td>
<td>TCCA STC O-LSA09/040 D, issue 3, dated 02 April 2012</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>25/01/2018</td>
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<td>10029931</td>
<td>EUROCOPTER UK LTD</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBITY</td>
<td>MBB-BK117 C-2</td>
<td>Cabin Configuration - Installation of Pre-Flight/Post-Flight System/Role Equipment</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Flight Manual Supplement No. 16, Issue No. 1, Design Report DRF-003 Rev. 1: Transfer from SAAB AB, NYKÖPING to SAAB AKTIEBOLAG. Rev. 2: Correction of DOA reference.</td>
<td>None</td>
<td>11/06/2010</td>
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<td>10029935</td>
<td>SAAB AB</td>
<td>REV. 2</td>
<td>MBB-BK117 C-2</td>
<td>Installation Provision/Video Signal in Mitsubishi MU-2B near pressurebulkheadRev. 1: Transfer from SAAB AB, NYKÖPING to SAAB AKTIEBOLAG. Rev. 2: Correction of DOA reference.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements - The following paragraph(s) at a later amendment: CS-23 Amdt. 1, dated 12 February 2009. The requirements for environmental protection and the associated certif...</td>
<td>TDM10-03-LEP Revision A dated 2010-04-30.</td>
<td>Target Towing provisions according to Swedish CAA approval CBA/71 must be installed. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft.</td>
<td>07/12/2023</td>
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<td>10029938</td>
<td>NSE INDUSTRIES S.A.</td>
<td>REV. 4</td>
<td>ASS50 B, ASS50 B1, ASS50 B3, ASS50 BA, ASS50 BB</td>
<td>Installation of DTSV2 Data Transmission SystemNote. Rev. 4 of this STC covers DTSV2 system upgrade due to component obsolescence, software update, small hardware changes and revision to STC Date List SDL-DTSV2-ASS350.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-27 Amdt 2, paragraphs 27.28, 27.39, 27.30, 27.39, 27.30, 27.39, 27.30, 27.39, 27.30, 27.39, 27.30, 27.39, 27.30.</td>
<td>Rotorcraft Flight Manual Supplement for DTSV2 Data Transmission System.doc. RMS- DTSV2-AS350 Rev. 00 dated 28 October 2013; later revisions of the above listed documents approved by EASA STC Data List: SDL-DTSV2-ASS350 Rev 01 dated 02/09/14.</td>
<td>Modification not approved for aircraft equipped with Weather Radar or AFCS (Automatic Flight Control System) Modification not approved for AS350B3 model helicopters without modification OP-4305 (Aerial 20 engine installation) incorporated. The installation of this modification by third persons is subject to...</td>
<td>29/09/2014</td>
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<td>10029937</td>
<td>S-TEC CORPORATION</td>
<td>REV. 4</td>
<td>FAA STC SA81040AC-D</td>
<td>Installation of S-TEC System SS/SSX Two Axis Automatic Flight Guidance System, Model ST-621 (14 Volt System)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAA Approved S-TEC Bulletin No. 721, dated 18- July-01. Revision 3, Older revision FAA Approved S-TEC Master Drawing List No. 92899, dated 18- July-01 Revision C, or later revision FAA Approved Supplemental Flight Manual, PN 891254, Revision 2, dated 18- July-01 or later approved revision in progress.</td>
<td>None</td>
<td>04/05/2010</td>
<td>Active</td>
<td>10029937</td>
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<td>10029942</td>
<td>DRF STIFTUNG LUFTRETTLUNGS-EINHEIT ZIEGE AG</td>
<td>REV. 3</td>
<td>MBB-BK117 B-2, MBB-BK117 C-1</td>
<td>Installation of Gamii GNS 530 FMS COMMNAV System and TCAS-1 Sky 8997CAS</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CS29 Amdt. 2: (not certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Approved S-TEC Bulletin No. 721, dated 18- July-01. Revision 3, Older revision FAA Approved S-TEC Master Drawing List No. 92899, dated 18- July-01 Revision C, or later revision FAA Approved Supplemental Flight Manual, PN 891254, Revision 2, dated 18- July-01 or later approved revision in progress.</td>
<td>None</td>
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<td>10029947</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td></td>
<td></td>
<td>CESSNA 340, 340A</td>
<td>Installation of Tornado Air Data Display/Altimeter and Avionik Straubing Altitude Preset System APS4A</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to thos certificates/approval: CS-23, Am.17, (the certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval</td>
<td>-Instructions for Installation: ASR-2009-025-F10-10-00, Rev. 0-Fighter Manual Supplements ASR-2009-025-FMS-05-00, Rev. 0-Instructions for Continued Airworthiness: ASR-2009-025-ICA-11-00, Rev.0 for later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>06/05/2010</td>
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<td>10029953</td>
<td>JET AVIATION AG, BASEL</td>
<td>REV. 1</td>
<td></td>
<td>B737-800</td>
<td>VIP Cabin Interior Outfitting</td>
<td>The Certification Basis for the original product remains applicable to thos certificates/approval. The certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval</td>
<td>Jet Aviation Documents: Master Data List: BB,J37545-001 Rev D dated 09, July 2010- AFM Supplement BB,J37545-201 Rev. E-Instr. for Continued Airworthiness; BB,J37545-150-01 Rev. for later revisions of the above listed documents approved by EASA</td>
<td>Limited to: S/N 37545</td>
<td>12/07/2010</td>
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<td>10029956</td>
<td>AIR CRUISERS COMPANY</td>
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<td>FAA STC SR20613NY</td>
<td>BELL 206 L-4</td>
<td>Bell 206L-4 Emergency Float Kit P/N 67585-101</td>
<td>The Certification Basis for the original product remains applicable to thos certificates/approval. The certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval</td>
<td>-EDN0258 revision IR dated 22 December 2008 - Rotorcraft Flight Manual/Supplement to later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>06/05/2010</td>
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<td>10029957</td>
<td>BOMBARDIER AEROSPACE</td>
<td></td>
<td>TCCA STC SA10-38</td>
<td>CL-600-2B16</td>
<td>Installation, Quick Access Recorder (QAR)</td>
<td>EASA TCDS No. M-A.023: The Certification Basis for the original product remains applicable to thos certificates/approval. The certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval</td>
<td>-Transport Canada STC No. SA10-38 Issue 1 dated 19 April 2010 - ACS-NAI Ltd. Modification Data Summary 051000 Rev N/C dated 19 April 2010, or later approved revision for Instructions for Continued Airworthiness. Refer to Bombardier document: TC-3130001-I-CA or later revisions of the above listed document</td>
<td>None</td>
<td>06/06/2010</td>
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<td>10029964</td>
<td>BVE AEROSPACE, Inc. - FSI</td>
<td>REV. 1</td>
<td>FAA ST02600SE</td>
<td>747-400</td>
<td>Amendment of STC 10029964 in order to remove EWS Limitations and update the MDL reference</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to thos certificates/approval: CS-23, Special Condition H-01, EWS83, AWPS2, Pneumatic Activation System</td>
<td>FAA STC No. ST02600SE dated 23 April 2010/FAA SOC dated 03 May 2010/Flight Structure Inc. Master Drawing List AFS7572-D02-010 Revision H dated September 22, 2010The Supplemental Instruction for Continued Airworthiness FAA Document AFS7572-D15 Revision C dated August 26, 2010 or later revisions of the above</td>
<td>None</td>
<td>04/10/2010</td>
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<td>10029965</td>
<td>AVIATION TRADERS LIMITED/EUROPE S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>Boeing 757-200 757-200</td>
<td>Boeing 757-200 229V Passenger ConfigurationRev. 1: Transfer STC</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI H-01/ Special Condition H-01 &quot;Enhanced Airworthiness Programme for Airlane Systems - ICA on EWIS&quot;</td>
<td>-ATL Modification Summary Sheet, Doc. No. MOD-11682-0001, Issue 3 dated 11th May 2010/ATL Master Document List, Doc. No.MOD-11682-0001, Issue 3 dated 30th May 2010 or later revisions of the above listed documents approved by EASA</td>
<td>As per associated Technical documentation for weight and balance/maximum allowable limits in the forward and aft cargo areas to be amended as per indicated in doc. N MOD-11682-0001, Issue 3 Limited to: MSN 275737 Prior to installation of this change/repair it must be determined that the interrelations...</td>
<td>10/11/2021</td>
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List of EASA Supplemental Type Certificates

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<td>10029967</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>FAA STC</td>
<td>STA0764SE-T</td>
<td>B747-400</td>
<td>First Class Cabin Upgrade BA 747-400 aircraft</td>
<td>EASA TCDS No. IM.A.1967</td>
<td>FAA STC number ST02003SE, dated 30 April 2010. FAA STC dated 03 May 2010.</td>
<td>Concurrent installation of EASA STC 10029964 is required. This modification may not be installed after 30 August 2010 unless (1) previously approved electrical wiring interconnected system (EWIS) instructions for continued airworthiness (ICA) have been evaluated and determined to be adequate for this inst...</td>
<td>07/05/2010</td>
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<td>10029999</td>
<td>ESG ELEKTRONIKSYS TEM-UND LOGISTIK GmbH</td>
<td>EASA R.S.01419</td>
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<td>Flight Manual Supplement No. 0005-FMS-025 Iss. 2or later revisions of the above listed documents approved by EASA-Means of Compliance Certification Program 0005-MOD-005 Iss. 2-Master Drawing List 0005-MDL-006 Iss. 3. Installation Instructions 0005-EO-024</td>
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<td>-</td>
<td>Restricted to ASN 9339 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced TCDS 2. WLAN system is compatible only to IEEE 802.11(b).</td>
<td>10/05/2010</td>
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<td>10030001</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC</td>
<td>SA04-113</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9339</td>
<td>EASA IM.A.009</td>
<td>Transport Canada C0C reference C-15-007 dated 04 May 2010 -TCCA STC No. SA04-113 Issue 3 dated 28 October 2004 - Aero Consulting Services Configuration Definition List D654000 RevisionEJ dated 23 April 2010 - Instructions for Continued Airworthiness GC93390000-IFCA.or later revisions of the above Iss...</td>
<td>Installation Instructions 0005-MDL-006. Iss. 1</td>
<td>10/06/2010</td>
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<td>10030002</td>
<td>BEECHCRAFT CORPORATION</td>
<td>FAA STC</td>
<td>SA109668C</td>
<td>KING AIR C90GTi</td>
<td>Installation of an upgraded Flight Management System (Version 4.0). This upgrade will offer a standard Flight Management Computer and will include an option for Search and Rescue Mission Profiles.</td>
<td>FAR 23 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Installation in accordance with in accordance with master drawing list338-00-0001, revision IR, dated July 27, 2010 or later revisions. Operation in accordance with Airplane Flight Manual Supplement for Collins FMS-3000 LPV Upgrade, P/N 338-00-014 dated February 2010 and Aircraft Flap...</td>
<td>None.</td>
<td>10/06/2010</td>
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<td>10030003</td>
<td>BEECHCRAFT CORPORATION</td>
<td>FAA STC</td>
<td>SA10970SC</td>
<td>KING AIR B200 AND B300</td>
<td>Installation of an upgraded Flight Management System (Version 4.0). This upgrade will offer a standard Flight Management Computer and will include an option for Search and Rescue Mission Profiles.</td>
<td>FAR 23 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Installation in accordance with in accordance with master drawing list337-00-0011, revision IR, dated January 27, 2010 or later revisions. Operation in accordance with Aircraft Flight Manual Supplement for Collins FMS-3000 LPV Upgrade, P/N 337-00-014 dated February 2010 and Aircraft Flight Manual...</td>
<td>None.</td>
<td>10/06/2010</td>
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<td>10030005</td>
<td>CL-600-2B16</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>CL-600-1A11 / -2A12 / -2B16</td>
<td>EASA AD 2009-007 and AD 2009-0173 for Transponder TDR-94D</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>None</td>
<td>None</td>
<td>17/05/2010</td>
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<td>10030005</td>
<td>CL-600-2B16</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>CL-600-1A11 / -2A12 / -2B16</td>
<td>EASA AD 2009-007 and AD 2009-0173 for Transponder TDR-94D</td>
<td>EASA IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>None</td>
<td>None</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>B757-300</td>
<td>EASA AD 2009-007 and AD 2009-0173 for Transponder TDR-94D</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>None</td>
<td>None</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>B757-300</td>
<td>EASA AD 2009-007 and AD 2009-0173 for Transponder TDR-94D</td>
<td>EASA IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>None</td>
<td>None</td>
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<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>None</td>
<td>None</td>
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<td>B757-300</td>
<td>EASA AD 2009-007 and AD 2009-0173 for Transponder TDR-94D</td>
<td>EASA IM.A.023 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>None</td>
<td>None</td>
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<td>West Pacific Air, LLC</td>
<td>FAA STC SA011566E</td>
<td>B36TC B36</td>
<td>Installation of a Pratt and Whitney PT6A-21 engine flat rated to 500Shp 2,200 RPM, and Harbeck modular HC-64N-3N/32B and HC-64N-3N/329 (B,K)-2 propellers. This certificate has been first issued on 17 May 2015 but has been re-issued for administrative reasons due to the loss of the original.</td>
<td>The Certification Basis for the original product remains applicable totsis certificate/approval. This certificate/approval involves a change to the requirements of fire prevention protection and/or a change to the certified noise and/or emissions levels.</td>
<td>Installation in accordance with Rockwell Engineering Master Drawing List450.00.000, Revision 5, dated June 30, 2007, or later approved revisions. Operation in accordance with Turbomeca/West Pacific Air LLC Flight Manual Supplement ref 460.1001, Change 8, dated July 16, 2007, or later approved revisions.</td>
<td>Model 826TC Only (The MT propeller option on the original FAA STC is not EASA approved at this time. In order to install the MT propeller compliance with EU noise and environmental standards must be shown.</td>
<td>24/04/2012</td>
<td>Active</td>
<td>10030059</td>
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<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 2</td>
<td>ATR 72-212, ATR 72-212A, ATR 72-211, ATR 72-201, ATR 42-500, ATR 72-101, ATR 42-300, ATR 42-320</td>
<td>Installation of single HF-600 Communication Transceiver System in ATR42 and ATR 72/212. The STC installs an additional HF communication device. This revision has added some ATR-42 aircraft models within the initial STC applicability.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (i) H-01. Enhanced Airworthiness Programme for Aircraft Systems – ICA on EWIS/GCS 25, AmI.25.21, 25, 27, 301, 303 to 307, 561, 571, 581, 25, 65011...</td>
<td>-Master Document List, ref: ATR72-212-02156 Revision 7 dated 16 April 2013. Aircraft Flight Manual Supplement, ref: ATR72-NY-02155-08 Revision 5 dated 09 August 2012. Instructions for Continued Airworthiness, ref: ATR72-NY-02155-07(Revision 6 dated 20 November 2012 - Weight &amp; Balance Supplement, ref: ATR72...</td>
<td>ATR Modification 2028 (Fuselage: Provide basic provisions for HF antenna/strap insulation) and Mod ATR Modification 3138 (Communications-HF system) installation of basic antenna must be embodied on the aircraft at the time of the STC installation. Prior to installation of this design change it must be done...</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10030066</td>
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<tr>
<td>HAWKER BEECHCRAFT CORPORATION (HBC)</td>
<td>FAA STC SA01566W/D</td>
<td>KING AIR B200 SERIES</td>
<td>Installation of modified Rockwell Collins Flight Guidance Computers (FGC-3000 units) as a solution to eliminate nuisance pitch trim wheel movement. FAR 23. The Certification Basis for the original product remains applicable to its certificate/approval. The certificate/ approval involves a change to the requirements of fire prevention protection and/or a change to the certified noise and/or emissions levels of the original product configuration unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Instructions for Replacement of FGC-3000 in Hawker Beechcraft Corporation King Air Models B200/B200C/B200GT/B200CGT, Models B200/B200C/B200GT/B200CGT, in accordance with Rockwell Collins Inc Master Drawing List ref PNRC/0241, Revision 1, dated September 15, 2009, Instructions for Initial and Continued Airworthiness, ref...</td>
<td>Prior approved installation of a Rockwell Collins Inc FGC-3000 Flight Guidance Computer Module (OPN 922-1108-031) is a prerequisite to this STC.</td>
<td>19/05/2010</td>
<td>Active</td>
<td>10030074</td>
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<td>BOMBARDIER INC.</td>
<td>TCCA STC C-LSA02-396/D</td>
<td>BD-700-1A10</td>
<td>Modification, 13 Passenger Interior Configuration</td>
<td>EASA 100309.00/000/00/00 The Certification Basis for the original product remains applicable to its certificate/approval. The certificate/approval involves a change to the requirements of fire prevention protection and/or a change to the certified noise and/or emissions levels of the original product configuration unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. C-LSA02-396/D Issue 1 dated 14 November 2002 TCCA SOC ref. C-10-0389 dated 17 May 2010. Aero Consulting Services (ACS) Modification Data Summary D1/2500 Revision 0/0. The following ACS Flight Manual Supplement &quot;Conversation: 13 Passenger Configuration &quot;GFM2500052 Rev N/C or later...</td>
<td>Restricted to ASN 9102 1. Previous Embodiment of EASA STC No. 10030097 &quot;Aircraft Replacement, IP Passenger Interior Configuration&quot; is required. 2. Bombardier Service Bulletin 700-31.018 Modification - Integrated Avionics Computers – Aironix 20017 Batch 1 IAC. Upgrade must be embodied.</td>
<td>19/05/2010</td>
<td>Active</td>
<td>10030078</td>
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<td>BOMBARDIER INC.</td>
<td>TCCA STC C-LSA02-332/D</td>
<td>BD-700-1A10</td>
<td>Aircraft Completion, 10 Passenger Interior Configuration</td>
<td>EASA 100309.00/000/00/00 The Certification Basis for the original product remains applicable to its certificate/approval. The certificate/approval involves a change to the requirements of fire prevention protection and/or a change to the certified noise and/or emissions levels of the original product configuration unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. C-LSA02-332/D Issue 1 dated 24 October 2002 TCCA SOC ref. C-10-0389 dated 14 May 2010. Aero Consulting Services (ACS) Modification Data Summary D1/2000 Revision 0 or later approved revision. The following ACS Flight Manual Supplements are required: &quot;G-FM24000033 Rev N/C Cabin Master Power...</td>
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<td>19/05/2010</td>
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<td>BOMBARDIER INC.</td>
<td>1A10</td>
<td>TC99-STC-SA04-113</td>
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<td>LBA STC 0060546B</td>
<td>SEE EASA APPROVED</td>
<td>REV. 3</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9331</td>
<td>-</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval. The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Restricted to ASN 9331 and. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS 2. WLAN system is compatible only to IEEE 802.11(b).</td>
<td>20/01/2022</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 3</td>
<td>LBA STC 0059546B</td>
<td>SEE EASA APPROVED</td>
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<td>REV. 4</td>
<td>LBA STC 0059546B</td>
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<td>MODEL LIST</td>
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<td>PIAGGIO AMERICA, Inc.</td>
<td>FAA STC SA60955SC</td>
<td>FAA STC SA10955SC</td>
<td>FAA STC SA10955SC &quot;Installation of a passenger seating system to replace the existing Decrane single seats with Modex FS1101 single seats&quot;</td>
<td>W &amp; B Manual update Doc. AN-1802-WB-01 Rev. 1</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval. The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>20/01/2022</td>
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<td>FEDERATION FRANCAISE DE VOL A VOILE (FFVV)</td>
<td>ASW02F</td>
<td>SLC-STC-M-0109-A01</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9331</td>
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<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval. The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
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<td>AEROSPACE DESIGN AND CERTIFICATION LTD.</td>
<td>BAE 146-200</td>
<td>New Cabin Configuration</td>
<td>JAR 25 Change 10The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Certification Program JN233ADC-1(2) Issue 3</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval. The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>21/06/2010</td>
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<td>MECAER AVIATION GROUP S.p.A.</td>
<td>REV. 1</td>
<td>AW139 VIP 305 INTERIORS P/N 6AB1AD015-001, Rev. J</td>
<td>Cab 139, AW139</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated certificate and or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>RSSM Supplement VIP 305 6AB1RFM-1-11, Rev.</td>
<td>Prior to installation of this design change it must be determined that the installation of the described modification will not have a detrimental effect on the airworthiness of the product.</td>
<td>28/10/2015</td>
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<td>10030099</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CO RPRATION AG (AIEC)</td>
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<td>MBB-BK 117 C-1</td>
<td>Cabin Interior Retrofit Installation on B117C-1</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>MBB-BK117 C-1 Rotorcraft Flight Manual Supplement - Doc. 854-075 Rev. B, dated 19.05.2010/Certification Program - Doc. 854-080 Rev. B dated 19.05.2010/Master Data List - Doc. 854-049...</td>
<td>21/05/2010</td>
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<td>AERO CONSULTING SERVICES LTD.</td>
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<td>TCCA STC NO. C-LSA10-048/D</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>EASA IM.A.009/06 - The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>TCCA STC No. C-LSA10-048/D Issue 1 dated 07 May 2010/TCCA Stc ref. C-10-0416 dated 19 May 2010/ACS-NAI Ltd. Modification Data Summary G568000 Revision NC or later/Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior, G568091 Revision NC...</td>
<td>Restricted to ASN 9319</td>
<td>21/05/2010</td>
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<td>10030118</td>
<td>AMIALE DE VOLTIGE AERIENNE(AVA)</td>
<td>REV. 1</td>
<td>CAP 232</td>
<td>Evolution of of CAP 232 to CAP 332 SC</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement CAP 332 SC Airplane Maintenance Manual Supplement CAP 332 SC or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>28/07/2011</td>
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<td>10030130</td>
<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U.</td>
<td>REV. 1</td>
<td>412EP</td>
<td>Installation of Weather Radar Bendix-King model RDR1400c inBell 412EP Helicopters Rev. 1: Transfer of STC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>INAER.210-1MD, Rev. 0 (Master Document List) INAER.210-06, Rev. 0 (Engineering Bulletin) INAER.210-5MS, Rev. 0 (Supplemental RPM) INAER.210-ICA, Rev. 0 (Instructions for Continued Airworthiness) or later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>01/02/2019</td>
<td>Active</td>
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<td>10030130</td>
<td>SAFRAN CABIN INC.</td>
<td>REV. 1</td>
<td>FAA STC ST01986LA-D</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Validation of FAA STC ST01986LA-D: Adjustable Cargo Net Installation Revision 1 is an administrative re-issuance to reflect the name change from C &amp; D to 20DIAC, Inc to SAFRAN CABIN, INC.</td>
<td>The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval without any impact to the noise database. For the airworthiness requirements, see attached EASA Approved Model List.</td>
<td>-</td>
<td>15/12/2023</td>
<td>Active</td>
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<td>10030139</td>
<td>AERO CONSULTING SERVICES LTD.</td>
<td>TD-700-1A11</td>
<td>TCCA STC NO. C-LSA10-041/D</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>EASA M.A. 009: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CRI-F-01 “Installation of Wireless System” The certificate, noise level or emissions levels of the original product remain unchanged...</td>
<td>TCCA STC No. C-LSA10-041/D Issue 1 dated 07 May 2010: TCCA S&amp;O ref. C-10-0427 dated 21 May 2010: ACS-N44 Ltd. Modification Data Summary G383000 Revision NC:Savannah Air Center Airplane Flight Manual Supplement for the Complete Custom Airplane Interior, G383090 Revision NC:Savannah Air Center Airplane...</td>
<td>Restricted to ASB 9324</td>
<td>27/05/2010</td>
<td>Active</td>
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<td>10030143</td>
<td>NSE</td>
<td>REV. 2</td>
<td>AS 355 N, AS 355 NP AS 355 F1, AS 355 F2</td>
<td>Installation of DTSv2 Data Transmission System Revision 1: The DTSv2 system is described in document IM-DTSv2-A355 and is defined in documents OD-411683 and DD-KIT-DTS-A355 Revision 2: Replacement of obsolete components, software update, small hardware changes and revision of STC Data List SDL-DTS...</td>
<td>The Certification Basis for the original product as amended by the following or an alternative airworthiness requirements: The following paragraph(s): at a later amendment: FAR 27 FAR 27.29 Am(i) 27.14... FAR 27.301 Original issue/FAR 27.305 Original issue/FAR 27.307...</td>
<td>Definition Dossier OD-411683 Revision 2 dated 20 June 2014: Certification Plan OERT-DTSV2-A355 Revision 1 dated 25 March 2020: Qualification Tests Result QTR-411683 Revision 0, Amdt D dated 06 September 2014: TCT Data List SDL-DTSv2-A355 Revision 1 dated 25 January 2020: The above list...</td>
<td>This change is applicable to all MSN except MSN equipped with weather radar and except MSN equipped with Automatic Flight Control System (AFCS). The approval holder shall fulfill the obligations of Part 21, Part 21, Part 21.</td>
<td>17/06/2020</td>
<td>Active</td>
<td>10030143</td>
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<td>10030147</td>
<td>AIRBUS S.A.S. (EASA.AS.03949)</td>
<td>AS300-200</td>
<td>Airbus Corporate Jet Centre Modification CJ0031 Issue 1 - VIP KIT - RAPF - Project</td>
<td>In accordance with the EASA TCDS A-064 plus: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: Certification Review Item CJ0031 Issue 1 &lt; Sheet ref. CJ0031 Issue 1&lt;</td>
<td>In accordance with the EASA TCDS A-064 plus: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: Certification Review Item CJ0031 Issue 1&lt;</td>
<td>1. Modification Approval Sheet ref. CJ0031 Issue 1&lt; (e.g. dated 27th May 2010): AFM Supplement Ref. FM-S MD0 CJ0031, dated May 01/10 or later revisions of the above listed document(s) approved by EASA.</td>
<td>This STC is limited to Airbus A320-200 MSN 3723 &amp; MSN 4117</td>
<td>28/05/2010</td>
<td>Active</td>
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### Notes
- The list includes STCs approved by EASA for various aircraft modifications and installations.
- Each STC is detailed with its associated certification basis, associated technical documentation, and limitations/conditions.
- The STCs are categorized by their type, including installation of weather radar, modifications to aircraft interiors, and other airworthiness requirements.
- The list also includes references to associated documents and modifications.
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<th>EASA STC no.</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
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<td>10030156</td>
<td>HARTZELL PROPPELLER INC.</td>
<td>FAA STD SAD2877CH-D</td>
<td>PA-46-350P, PA-46R-350T</td>
<td>EASA validation of FAA STC SA02877CH-Installation of Hartzell Propellers H3YR-1N/1N7605(II)+2; H33YR-1N/1N7605(II)+2</td>
<td>CS-23.8 NSC</td>
<td>Field Installation Instructions:1. INST_060808_F2FLdfvr PA-46-350P, SN 4622001 to 4622200, 4636001 to 4636132, 1. INST_080808_F2FLdfvr PA-46-350P, SN 4636132 and up, PA-46R-350T, S/N 4692001 and upFactory Installation Instructions:1. INST_060808_F2FLdfvr PA-46-350P, SN 4636132 and up, PA-46R-350T...</td>
<td>For further limitations see applicable associated technical documentation</td>
<td>31/05/2010</td>
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<td>10030161</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>2. REIMS CESSNA 337</td>
<td>Introduction of Bubble Windows</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>STC Twenty One Limited Master Document List No. S21.MDL-0752 at Initial Issue dated 26 May 2011 for later revisions of the above listed documents approved by EASA</td>
<td>Restricted to un-pressurized versions of the Cessna/Cessna Reims 337 aircraft</td>
<td>31/05/2010</td>
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<td>10030166</td>
<td>AIR COMM CORPORATION LLC</td>
<td>FAA STC SR00463DE</td>
<td>A119, ANY119 MKII</td>
<td>FAA STC No. SR00463DE Installation of Air Conditioning System</td>
<td>- FM Supplement Doc. No. A119-1, Rev. 6, 11 June 2009 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) - FAA STC No. SR00463DE amended 27 November 2007-Air Comm Corp. Master Drawing List Report No. D...</td>
<td>None</td>
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<td>10030206</td>
<td>AEROSPACE DESIGN AND ENGINEERINGCONSULTANTS Ltd.</td>
<td>FAA STD ST108295E</td>
<td>747-400</td>
<td>Reconfiguration of compartment zone A in accordance with Amendment Modification summary MOD-0388 Rev. B dated 29 March 2010 or laterapproved revision. This STC validates FAA STC ST108295E</td>
<td>-Jamco America Inc. Instructions for Continued Airworthiness, document number JD-09AC03 Rev. New, dated March 06, 2008for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>None</td>
<td>02/06/2010</td>
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<td>10030208</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>A318-112</td>
<td>Installation of a VIP cabin for a maximum of 19 passengers. The design details are described in General Description C- AAA057-SPA Delletion of the Airworthiness Limitation Item reference 530000-01-5 for A318-112 Elite 14 MSN 4189</td>
<td>The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this certificate/approval: CRI D-1 Width of Aisle for EASA.A.S.03549</td>
<td>Jamco America Inc. Instructions for Continued Airworthiness, document number JD-09AC03 Rev. New, dated March 06, 2008for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>None</td>
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<td>STC No.</td>
<td>Company</td>
<td>Model</td>
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<td>Certification Basis</td>
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<td>Certification Condition</td>
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<td>10030228</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A380/841-842</td>
<td>Installation of Class 2 EFB Laptop Cradle</td>
<td>Certification basis</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the Certificate holder per TCDS EASA A.4.101 and Certification review item CRI F-37. The requirements for environmental protection and the associated certified noise emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10030229</td>
<td>STC TWENTY ONE LIMITED</td>
<td>B767-200/-300</td>
<td>Introduction of Air Conditioning Filter</td>
<td>Certification basis</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the Certificate holder per FAA TCDS 24NM and assessed by CRI D-01 and CRI H-01. The requirements for environmental protection and the associated certified noise emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10030237</td>
<td>TÄBY AIRMAINTENANCE E AKTIEBOLAG</td>
<td>SAAB SF340A, SAAB 340B</td>
<td>Rework of the Class E Compartment for inclusion of a chair seating area for up to 5 occupants, S21.25-63-1355 Rev. 1</td>
<td>Certification basis</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the Certificate holder per FAA TCDS 24NM and assessed by CRI D-01 and CRI H-01. The requirements for environmental protection and the associated certified noise emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10030239</td>
<td>BOMBARDIER INC.</td>
<td>TCJA STC SA05-60</td>
<td>Installation, Interior Completions.</td>
<td>Certification basis</td>
<td>EASA IM.A.006 The Certification Basis for the original product remains applicable to the certificate/approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10030240</td>
<td>BOMBARDIER INC.</td>
<td>TCJA STC SA05-60</td>
<td>Installation, Interior Completions.</td>
<td>Certification basis</td>
<td>EASA IM.A.006 The Certification Basis for the original product remains applicable to the certificate/approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>AIR SUPPORT INTERNATIONAL S.r.l.</td>
<td>PA-30</td>
<td>Installation of Garmin GNS 430W approved for B-RNAV and other avionic equipment</td>
<td>Certification basis</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval. The applicant elected to comply with CS-23 par. 1301, 1309, 1307, 1321, 1331, 1332, 1339, 1351, 1353, 1357, 1360, 1367, 1381, 1529; EASA AMC 26-4; EASA Circular 2004/02/CF (or subsequent revisions off).</td>
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<td>List of EASA Supplemental Type Certificates</td>
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<td><strong>Rev. Level</strong></td>
<td><strong>Foreign/NASTC ref.</strong></td>
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<td><strong>Description</strong></td>
<td><strong>Certification basis</strong></td>
<td><strong>Associated Technical Documentation</strong></td>
<td><strong>Limitations/Conditions</strong></td>
<td><strong>Date of Issue</strong></td>
<td><strong>Status</strong></td>
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<td>ESG ELEKTRONIKSYS TEM-UND LOGISTIK-GmbH</td>
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<td>WIPARE Inc.</td>
<td>FAA STC SA0059WI</td>
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<td>Installation of the Wipaire Single Point Fueling System on Cessna 208i.a.w. FAA STC SA0059WI</td>
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<td>MECERA AVIATION GROUP S.p.A.</td>
<td>AB139/AW139</td>
<td>REV. 1</td>
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<td>Cabin Configuration VIP-A P/N 6-A202/01/010-003. This STC supersedes EASA STC EASA R.S.00484.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>EASA STC EASA.R.S.00485.</td>
<td>Approval of this change in type design applies only to those AB139rotorcraft previously equipped with EASA approved installation of AgustaKi Vip.</td>
<td>09/06/2010</td>
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<td>MECERA AVIATION GROUP S.p.A.</td>
<td>AB139/AW139</td>
<td>REV. 1</td>
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<td>Cabin Configuration VIP-B P/N 6-A202/01/010-001. This STC supersedes EASA STC EASA R.S.00485.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval. The requirement for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>EASA STC EASA.R.S.00484.</td>
<td>Approval of this change in type design applies only to those AB139rotorcraft previously equipped with EASA approved installation of AgustaKi Vip.</td>
<td>09/06/2010</td>
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<td>EUROCOPTER UK Ltd.</td>
<td>MBB-BK 117 C-2</td>
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<td>Wescam HD FLIR camera system. Installation of Wescam HD-3 FLIR camera system with build standards for electromagnetic installation of MX-15. Analogic system in accordance with EC UKchange no. MHU/145/11, Master Drawings A31451211-100-18.2, Thermal installation is including FLIR turn, hand controller, a GPS A...</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval.</td>
<td>Eurocopter UK BK117 C-2 (EC145) Supplement No. 18. Issue for later revisions of the above listed documents approved by EASA-Instructions for Continued Airworthiness -IAMI/145/1211-001, Iss. A-Design Report DR 1451211-001, Iss. A</td>
<td>None.</td>
<td>Improvements made to the EASA STC for installation on AW139 rotorcraft fleet approved by EASA-Instruction no. 145. Revision 4.</td>
<td>09/06/2010</td>
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<td>GAIA AVIATION (ENGINEERING) LIMITED</td>
<td>10030289</td>
<td>AVIDYNE CORPORATION</td>
<td>10030294</td>
<td>DINAN AVIATION</td>
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<td>FAA STC 6-A03RFM-1-6. Rev. 0</td>
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List of EASA Supplemental Type Certificates
easa.europa.eu

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<th>STC Holder</th>
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<th>Limitations/Conditions</th>
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<td>MECAIR AVIATION GROUP S.p.A.</td>
<td>REV. 1</td>
<td>AB139/ AW139</td>
<td>Cabin Configuration VIP-1 P/N 6-1030/50101-001</td>
<td>This STC superseded EASA STC EASA R.S.00487.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise/emissions of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>SFM 6A/0200/P-M-7.</td>
<td>Approval of the change in type design applies only to those AB 139 rotorcraft previously equipped with EASA approved installation of Ag Way VIP Cabin interior (UAE) P/N 3G2S0F00111.</td>
<td>09/06/2010</td>
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<td>MECAIR AVIATION GROUP S.p.A.</td>
<td>REV. 2</td>
<td>AB139/ AW139</td>
<td>FVFD Bushhead P/N 6 AB1AD005-001</td>
<td>Installation This STC superseded EASA STC EASA R.S.01228.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise/emissions of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Owner's Manual P/N 6AB1WNNA-2.4.</td>
<td>None.</td>
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<td>EUROCOPTER UK Ltd.</td>
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<td>BK117 C-2 (EC145)</td>
<td>Dual Martin Baker Swivelling Passenger Seats</td>
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<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise/emissions of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>-ECUK RFMS Number 19 Issue 1 for later revisions approved by EASA DR/145/1196-002 Issue A</td>
<td>As identified in RFMS 19</td>
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<td>TECH-TOOL PLASTICS, Inc.</td>
<td>FAA STC</td>
<td>SR09127RC-D</td>
<td>AS350 B1, AS350 B2, AS350 B3, AS350 BA, AS350 D</td>
<td>Installation of a Vertical Sliding Window in the Right and/or Left Hand Sliding Door FAA STC SR09127RC-D</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- Drawing List 350A42-1750, Rev B, dated August 25, 2000 - Rotorcraft Flight Manual Supplement (FMS) 350A42-066, Revision B, dated July 24, 2008 - IA, Doc Number TTPICA-350-001, dated July 27, 1999 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 200...</td>
<td>None.</td>
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<td>REV. 2</td>
<td>AB139/ AW139</td>
<td>Liners Silva P/N 6AB10A3010-001/Rev. 1: Superseude of EASA STC EASA R.S. 01534 Rev. 2. 2: Introduction of trim panel system compatibility to AB139 helicopter sliding door configuration</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement AB1RFM-1.1 Rev. Eor later revisions of the above listed documents approved by EASA-NDC-6A16AD-241/11, Issue A Change Definition (Description, Configuration) NDC-6A16AD-241/2, Issue A Certification Program (CCL &amp; MOC) NDC-6A16AD-241/3, Issue A Change Approval (Master)</td>
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<td>AIR METHODS CORPORATION</td>
<td>FAA STC</td>
<td>SR00250DE-D</td>
<td>MD900</td>
<td>Installation of an EMS Interior in a MD902 Helicopter</td>
<td>As described in FAA TCDS H19WNThe Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List Report No. 231-000-001, revision J, dated March 5, 2010 - Rotorcraft Flight Manual Supplement &quot;EMS Interior&quot;, No. 080-0200 revision E, dated June 5, 2008 - Rotorcraft Flight Manual Supplement &quot;All Equipment Frame&quot;, No. 080-0200, revision A, dated May 9, 2007 - Instructions for Continue...</td>
<td>If damage or excess wear warrants structural repair or alteration to the rotor system, pre-approval authorized by the holder of this STC is required in accordance with AMMS 050-0167, Rev. (-), dated June 13, 1996 for later FAA accepted revisions. The multi-purpose floor system applies only to those AB 139 rotorcraft previously equipped.</td>
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<td>LUFTHANSA TECHNIK AG</td>
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<td>A340-500</td>
<td>A340-500 Cabin retrofit ET03D013 STC modifies the cabin layout by replacing First Class seats and all associated seat monuments.</td>
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<td>10030404</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>EC 120 B</td>
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<td>Panel Modification including position change of Apollo MX20 (Gallm3GMO200)-HSI Indicator K5 525 and installation of new Altitude Indicator and Moving Terrain MT-Vision MFD.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this STC/revision.</td>
<td>Definition and installation in accordance with Avionik Straubing Entwicklungs GmbH Master Documents List ASR-2010-005-MDL -01-008-MIL/00/10-Avionik Straubing GmbH Installation Instruction ASR-2010-005-IF -10-000-Inspection and Maintenance in accordance with Avionik Straubing GmbH Instructions for Continued Airworthiness. None</td>
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<td>10030405</td>
<td>BENZ AIRBORNE SYSTEMS</td>
<td>FAA STC SA19696CH</td>
<td>DHC-6-100, -200, -300</td>
<td>Installation of Torque Pressure System on DHC-6 in acc. to FAA STC SA19696CH</td>
<td>The Certification Basis for the original product remains applicable to this STC/revision.</td>
<td>Armstrong Aerospace Master Data List No. DC104-3021-00; Rev. B, dated July 18, 2005 BENZ Airborne Systems, Engine Torque Pressure Indicating System DHC-6-300, Aircraft Maintenance Manual Supplement; Rev. A, March 10, 2004; later revisions of the above listed documents approved by EASA.</td>
<td>15/06/2010</td>
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<td>10030407</td>
<td>TIMCO AVIATION SERVICES INC.</td>
<td>FAA STC ST0828AT-D</td>
<td>B767-300</td>
<td>Interior reconfiguration. B767-300 installation of FAA STC ST0828AT-D. The aircraft will have the following configuration changes: a) additional coach (Y) class seating, b) additional seat track installation in associated structural modifications, c) removal of a lavatory (MS-13) at the R2 door, installation...</td>
<td>As per EASA TCDS IMA A.035 for this STC the following CRI/Special Condition is applicable and of particular interest: &quot;Enhanced Airworthiness Programme for Airliner System - ICA on EWIS&quot;. The requirements for environmental protection and the associated certified noise and/or emissions remain applicable to this STC/revision.</td>
<td>-Master Data List No. 09T116R009, Rev. A, dated April 23, 2010; later revisions of the above listed documents approved by EASA. As per FAA STC ST0828AT-D, for subsequent revisions.</td>
<td>15/06/2010</td>
<td>Active</td>
<td>10030407</td>
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<tr>
<td>10030408</td>
<td>BENZ AIRBORNE SYSTEMS</td>
<td>FAA STC SA19707CH</td>
<td>DHC-6-100, -200, -300</td>
<td>Installation of Engine Oil Pressure Indicating System on DHC-6 in acc. to FAA STC SA19707CH</td>
<td>The Certification Basis for the original product remains applicable to this STC/revision.</td>
<td>Armstrong Aerospace Master Data List No. DC104-3020-00; Rev. B, dated July 18, 2005 BENZ Airborne Systems, Engine Oil Pressure Indicating System DHC-6-300, Aircraft Maintenance Manual Supplement; Rev. A, March 10, 2004; later revisions of the above listed documents approved by EASA.</td>
<td>15/06/2010</td>
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<td>10030408</td>
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<td>10030420</td>
<td>SCHEDEL ENGINEERING Ltd.</td>
<td>ASK 21, ASK 21 M</td>
<td></td>
<td>ASK 21 Front Instrument Panel Modification - Modification of the ASK 21 front seat instrument panel to allow more leg space for the pilot.</td>
<td>The Certification Basis for the original product remains applicable to this STC/revision.</td>
<td>Doc No 10_ASK21 FIP-001 None</td>
<td>15/06/2010</td>
<td>Active</td>
<td>10030420</td>
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<td>10030424</td>
<td>PETERSEN AVIATION, INC.</td>
<td>FAA STC SA0214WI</td>
<td>CESSNA F182P</td>
<td>Operation of airplane on unleaded automotive gasoline, 87 minimum research octane index and leaded automotive gasoline, 88 minimum anti-knock index (RON + MON) 2.0 per ASTM Specification D-439 in acc. to FAA STC SA0214WI.</td>
<td>The Certification Basis for the original product remains applicable to this STC/revision.</td>
<td>EASA approved AFMS for Reims-Cessna F182 P, Doc.No. 163P- SA0214WI, Issue 1; dated June 10, 2010; Petersen Drawing No. C71894 dated July 1, 1994, revised December 15, 1990; later revisions of the above listed documents approved by EASA. 1. Limited to those airplanes equipped with gravity feed carburettor and certified (original or STC) EASA approved engine installations. All engines must have separate approval for operation on automotive gasoline in the form of Supplemental or Type Certificate Approval 2. Alcohol/ethanol.</td>
<td>15/06/2010</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<td>WESTERN AVIONICS</td>
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<td>Coates Prime II Installation/TCGA STC SA03-42</td>
<td>FAR-25 as per EASA M.A. 1917The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable bolts certificate/ approval/Special Condition H-01 &quot;Enhanced Airworthiness Programme for Aeroplane/systems - ICA on EWIS&quot; as detailed in CRI...</td>
<td>Installation must be in accordance with Western Avionics Engineering Order 89598-401, Rev. 1, dated April 24, 2003, or later Transport Canada/approved revision. Maintenance must be in accordance with Western Avionics Maintenance/Manual Supplement, 89598-701, Rev. 1, accepted May 8, 2003, 89596-702 Rev. 8...</td>
<td>None</td>
<td>16/02/2010</td>
<td>Active</td>
<td>10030452</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>B200-200GT</td>
<td>Air Alpha STC Definition Document EO-B200-25-146: Flight Inspection/System Interface Installation King Air 200 Series.</td>
<td>The Certification Basis for the original product remains applicable bolts certificate/approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable. bolts certificate/approval.</td>
<td>Air Alpha Master Data List No. MSL EO-B200-25-145 Rev. 06 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>12/01/2016</td>
<td>Active</td>
<td>10030456</td>
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<tr>
<td>GAMA ENGINEERING LTD.</td>
<td>B010S</td>
<td>Mounting boom for camera installation.Certification of the installation of a mounting boom which can be useful for installation of a camera. The camera installation provisions and themaximum weight and frontal area of cameras which can be used with themounting boom are part of this STC. No special came...</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable bolts certificate/approval. The applicant has shown compliance with the current EASA certificationspecification CS-27 Amendment 2. The requirements for environmental...</td>
<td>Rotorcraft Flight Manual Supplement. RPM 1089-01, Issue 1, Instructions for Continued Airworthiness: ICA on EWIS polymers, as detailed in CRI...</td>
<td>CAT-A operation is prohibited (CAT-A operation is excluded from certification). Maximum weight for cameras used is 43.1 kg and maximum frontal area is 1470 sq.cm. The used camera has to be certified separately.</td>
<td>16/06/2010</td>
<td>Active</td>
<td>10030458</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>B200-200GT</td>
<td>Provision for Flight Inspection use of MLS navigation input.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/ certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Air Alpha A/S Master Data List Numbered: MDL EO-B200-25-129 Rev. 04 dated 04 June 2010; - Installation in accordance with the Accomplishment Instructions Ref.AEO-B200-25-129 Rev 01 dated 19 May 2010.</td>
<td>MINIMUM FLIGHT CREW: FIS MLS Navigation may only be used with two (2) flight crew members seated in the cockpit.FLS NOT FOR PRIMARY NAVIGATION. FLIGHT INSPECTION GUIDANCE ONLY. STC Holder approval required prior to application by any other party. Prior to installation of this modification it must be...</td>
<td>19/07/2016</td>
<td>Active</td>
<td>10030470</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>B200-200GT</td>
<td>MID-124 Direction Finder Antenna Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/ certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Air Alpha A/S Master Data List Numbered: MDL EO-B200-25-125 Rev. 08 dated 08 July 2009 - Instructions for Continued Airworthiness (i.e. AirAlpha An SMaintenace Manual Supplement Ref. ATPS 0212) Rev...</td>
<td>STC Holder approval required prior to application by any other party. Prior to installation of this modification it must be determined in flight. Interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect(s) on the airworthiness...</td>
<td>19/07/2016</td>
<td>Active</td>
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<td>BOND HELICOPTERS EUROPE LTD</td>
<td>CAA UK RESPONSIBILITY</td>
<td>EC135 T2(OPDS), EC135 T2+</td>
<td>Engine Usage Monitoring SystemInstallation of an Engine Usage Monitoring System in accordance with change number BOND/MOD/010 as defined in DL010-0699 to monitor certain airframe parameters, further described in Design Report DR010-001.</td>
<td>Flight Manual Supplement EC135 T2, T2+ &quot;Engine Usage Monitoring System&quot;, FMS0010-002, Issue 4 dated 08 July 2009 - Instructions for Continued Airworthiness (i.e. AirAlpha An SMaintenace Manual Supplement Ref. ATPS 0212) Rev...</td>
<td>None</td>
<td>17/06/2010</td>
<td>Active</td>
<td>10030474</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>STC Number</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10030479</td>
<td>EAGLE COPTERS USA, INC.</td>
<td>REv. 1</td>
<td>FAA STC SR002195E</td>
<td>2. A3355F2, A3355N 1. A3355E, A3355F, A3355F1 1. A3355B, A3355B1, A3355B2 1. A3355BA, A3356D</td>
<td>Horizontal Stabilizer Colour Camera Mount per MDL GA161 Notes 1. The STC was initially issued to Geneva Aviation, Inc. (initial reference: 10030479). Revision 1 to this STC reflects the transfer of 10030479 from Geneva Aviation, Inc. to Eagle Copters USA, Inc.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval</td>
<td>Master Drawing List GA161 Rev. - initial release dated 21/07/1995 revision N/C of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions after this decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>11/03/2015</td>
<td>Active</td>
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<tr>
<td>10030495</td>
<td>NORTHEAST AEROSPACE TECHNOLOGIES INC</td>
<td></td>
<td>A319-132</td>
<td></td>
<td>Interior Reconfiguration and In-Flight Entertainment (IFE) Installation on Ethad Airways A319-132 Aircraft</td>
<td>In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval Class H-01 Enhanced Airworthiness Programme forAeroplane Systems - ICA on EWI/S-Tne c</td>
<td>NAT Certification Summary, ref. doc. N0338-D00, Revision - dated 17th June 2010. Master Drawing List, ref. doc. N0338-S10, Revision C, dated 16th June 2010 later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>18/06/2010</td>
<td>Active</td>
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<td>10030503</td>
<td>APICAL INDUSTRIES, Inc.</td>
<td></td>
<td>FAA STC SR00259LA</td>
<td>BELL 412, BELL 412EP BELL 245B-1, BELL 205A-1 BELL 242, BELL 241B BELL 204B</td>
<td>Installation of Cable Cutters Kit in accordance with MasterDocument List MD42-101, Revision B, dated May 12, 2010 or later FAA approvedrevision.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval</td>
<td>Instructions for Continued Airworthiness ICA412-101, Revision N/C, datedJanuary 5, 2010 later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions after this decision)</td>
<td>None</td>
<td>21/06/2010</td>
<td>Active</td>
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<tr>
<td>10030504</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>Rev. 1</td>
<td>FAA STC SR01166SE</td>
<td>AS 350 B, AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA, AS 350 D</td>
<td>Revision of STC EASA.IM.R.S.00542 introducing additional cargo hook kit200-282-03.</td>
<td>As specified in TCDS R.0088 The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>-FAA STC ST002195E with Approved Model List. See AML page 11 of 13 for BD-100-1A10 Master Drawing List Number 226 for Vision Safe Corporation EVAS Model107STC-235 and 107STC-236 onBombardier Aerospace BD-100-1A10. Revision C dated 14th March 2007. BD-100-1A10 Airplane Flight Manual Supplement EVAS STC</td>
<td>USA, Inc. Transfer of 10030479 from Geneva Aviation, Inc. to Eagle Copters Services GmbH. Notes: 1. The STC is approved only for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. 2. The STC is approved only for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>21/06/2010</td>
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<td>STC Holder</td>
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<td>VH AEROSPACE Inc.</td>
<td>10030509</td>
<td>STC Holder</td>
<td>TCCA STC: SH09-54</td>
<td>Triple Stacking Litter System Installation/Validation of TCCA STC SH09-54 issue 1</td>
<td>None</td>
<td>21/06/2010</td>
<td>Active</td>
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<td>AVIATION SUPPORT GmbH</td>
<td>10030519</td>
<td>Certification basis</td>
<td>AS 350 BB, AS 350 D</td>
<td>The Certification Basis for the original product remains applicable to this certification/</td>
<td>Original, Revision 1</td>
<td>07/06/2021</td>
<td>Active</td>
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<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>10030532</td>
<td>Certification basis</td>
<td>AS 350 B, AS 350 B1, AS 350 B2</td>
<td>Cargo Mirror Set - Type Air Zermatt</td>
<td>Approval of this change applies only to Bell Helicopter models 206A and 206B previously equipped with Bell cargo suspension systems. Part Number(P/N) 206-072-900-1, or 206-072-900-101, or 206-072-900-103; Bell cargohook provision kit, P/N 206-706-335-3, 206-706-335-5, or206-706-335-105; and</td>
<td>28/02/2017</td>
<td>Active</td>
<td>10030532</td>
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<td>INFLIGHT CANADA Inc.</td>
<td>10030534</td>
<td>Certification basis</td>
<td>AS 350 B, AS 350 D</td>
<td>Audio Video on Demand (AVOD) and In-Seat Power Supply (SPS) System Installation</td>
<td>Approval of this change applies only to Bell Helicopter models 206A and 206B previously equipped with Bell cargo suspension systems. Part Number(P/N) 206-072-900-1, or 206-072-900-101, or 206-072-900-103; Bell cargohook provision kit, P/N 206-706-335-3, 206-706-335-5, or206-706-335-105; and</td>
<td>10/06/2012</td>
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<tr>
<td>AERODATA AG</td>
<td>10030547</td>
<td>Certification basis</td>
<td>AS 350 B, AS 350 D</td>
<td>Installation of Aerodata Flight Inspection System AD-AFIS-130</td>
<td>None</td>
<td>23/06/2010</td>
<td>Active</td>
<td>10030547</td>
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**Certification basis:**
- **AS 350 BB, AS 350 D:** The Certification Basis for the original product remains applicable to this certification/.
- **Cargo Mirror Set - Type Air Zermatt:** Approval of this change applies only to Bell Helicopter models 206A and 206B previously equipped with Bell cargo suspension systems. Part Number(P/N) 206-072-900-1, or 206-072-900-101, or 206-072-900-103; Bell cargohook provision kit, P/N 206-706-335-3, 206-706-335-5, or206-706-335-105; and.
- **Audio Video on Demand (AVOD) and In-Seat Power Supply (SPS) System Installation:** Approval of this change applies only to Bell Helicopter models 206A and 206B previously equipped with Bell cargo suspension systems. Part Number(P/N) 206-072-900-1, or 206-072-900-101, or 206-072-900-103; Bell cargohook provision kit, P/N 206-706-335-3, 206-706-335-5, or206-706-335-105; and.
- **Installation of Aerodata Flight Inspection System AD-AFIS-130:** None.

**Limitations/Conditions:**
- **Triple Stacking Litter System Installation/Validation of TCCA STC SH09-54 issue 1:** None.
- **Cargo Mirror Set - Type Air Zermatt:** Approval of this change applies only to Bell Helicopter models 206A and 206B previously equipped with Bell cargo suspension systems. Part Number(P/N) 206-072-900-1, or 206-072-900-101, or 206-072-900-103; Bell cargohook provision kit, P/N 206-706-335-3, 206-706-335-5, or206-706-335-105; and.
- **Audio Video on Demand (AVOD) and In-Seat Power Supply (SPS) System Installation:** Approval of this change applies only to Bell Helicopter models 206A and 206B previously equipped with Bell cargo suspension systems. Part Number(P/N) 206-072-900-1, or 206-072-900-101, or 206-072-900-103; Bell cargohook provision kit, P/N 206-706-335-3, 206-706-335-5, or206-706-335-105; and.
- **Installation of Aerodata Flight Inspection System AD-AFIS-130:** None.

**Date of Issue:**
- **10030509:** 21/06/2010.
- **10030519:** 07/06/2021.
- **10030532:** 28/02/2017.
- **10030534:** 10/06/2012.
- **10030547:** 23/06/2010.
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<tr>
<td>10030551</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>EC135 P2+</td>
<td>Installation of an Intrad Satellite Telephone on board</td>
<td>The Certification Basis for the original product remains applicable (to certificate/ approval) The certificated noise and or emissions levels of the original product unchanged and remain applicable to this certificate/approval</td>
<td>Master Drawing List Ref 13504-RPT04, Edition 2, dated 19/06/2010</td>
<td>Not Applicable</td>
<td>23/06/2010</td>
<td>Active</td>
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<td>10030620</td>
<td>AVIATION SERVICE A.S.</td>
<td>L410 UVP-E, L410 UVP-E9, L410 UVP-E20</td>
<td>TCA'S II CAS67A, Mode S transponder, MSTR67A, and Air Data Display Unit AD32 installation to comply with regulation requirements.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable (to certificate/approval) FAR Part 23, Amdt. 99</td>
<td>-FM34 40/01 001, Flight Manual Supplement CAS 67A and MSTR67A FM34 40/01 002, Doplník Letové příručky CAS 67A a MSTR67A FM34 40/01 001., Flight Manual Supplement AD 32 FM34 101 002 Doplník Letové příručky AD 32</td>
<td>None</td>
<td>24/06/2010</td>
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<td>10030614</td>
<td>DECA AVIATION ENGINEERING Ltd dba PAL AEROSPACE</td>
<td>TCCA STC O-LSA08-0010D</td>
<td>B737-800 Passenger Seating Arrangement - Installation / Relocation of passenger seats in 186-189 Pax Configuration and installation of Emergency Equipments</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable (to certificate/approval)</td>
<td>DECA Modification Summary No. MS05289, Rev. 16 dated 06 January 2010</td>
<td>Only valid for MSN 28373, 28375, 29351, 29642, 29683, 30637, 30671, 33003, 33029, 33869, 34704, 3593</td>
<td>25/06/2010</td>
<td>Active</td>
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<td>10030620</td>
<td>BOMBARDIER AEROSPACE</td>
<td>BD-100-1A10 (CHALLENGER 300)</td>
<td>Installation of Miniature Quick Access Recorder</td>
<td>Certification Basis as per EASA TCDS MA.280 issue 2, 8th October 2008plus The Certification Basis for the original product remains applicable (to certificate/approval) The certificated noise and or emissions levels of the original product unchanged and remain applicable to this certificate/approval</td>
<td>-Transport Canada TCDS MA.280 issue 1, dated 02/10/2010, Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary ref G259/000 dated 30 March 2010, Revision No.-Instructions for Continued Airworthiness ref. RAL-BY/100-3111-CA, Revision 01</td>
<td>None</td>
<td>28/06/2010</td>
<td>Active</td>
<td>10030620</td>
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<td>10030622</td>
<td>AEROSEVIS, S.R.O.</td>
<td>L 410 UVP</td>
<td>The set of the safety modifications in order to prepare the alternative in the compliance to the EASA AD 208-1003.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable (to certificate/approval) FAR Part 23, Amdt. 34, dated 17 February 1987 through Amdt. 49, dated 11 March 1996</td>
<td>-L 410 UVP Flight Manual Supplement No. STC AE/103-DFP1, dated March15/10/4, L 410 UVP Wiring Manual Supplement No. STC AE/103-DFP1, dated March10/10/L 410 UVP Maintenance Manual Supplement No. STC AE/103-DFP1, datedMarch19/10/L 410 UVP Maintenance Schedule Supplement No. STC AE/103-DFP1, datedMarch 29/1</td>
<td>None</td>
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<td>10030640</td>
<td>LATITUDE TECHNOLOGIES CORPORATION</td>
<td>TCCA STC SH10-16</td>
<td>AS335F2, AS335N, AS335NP, AS335S, AS335F, AS335F1, Installation of Latitude Technologies S100 Tracking System</td>
<td>The Certification Basis for the original product remains applicable (to certificate/approval) The requirements for environmental protection and the associated certificated noise and or emissions levels of the original product unchanged and remain applicable to this certificate/approval</td>
<td>1117-MD-001 revision N, C dated 7 April 2010 - Master Drawing List 1117-MD-004 revision 01 dated 7 April 2010 - Instructions for Continued Airworthiness</td>
<td>None</td>
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<td>LATITUDE TECHNOLOGIES CORPORATION</td>
<td>REV. 1</td>
<td>TGOA  STC: SH10-16</td>
<td>206A-3, 206L-4, 407-208B, 206L-1, 206L, 206L-1, 208, 208A-1, 208B</td>
<td>Installation of Latitude Technologies S100 Tracking System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval,</td>
<td>None</td>
<td>11/17-645,001 revision N-C dated 7 April 2010 - Master Drawing List: 1117-400-004 revision 01 dated 7 April 2010 - Instructions for Continued Airworthiness</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0371AT-D</td>
<td>GIV-X</td>
<td>Installation of an Audio Video Distribution System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval,</td>
<td>Installation in accordance with Gulfstream Index List GC415928091, Revision C, dated 10 September 2009 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004004CF (or subsequent revisions or this decision)</td>
<td>None</td>
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<td>SIERRA INDUSTRIES Ltd.</td>
<td>REV. 1</td>
<td>FAA STC SA3634SW</td>
<td>U206 LANDPLANES E, F, G</td>
<td>Auxiliary Fuel System: Installation of three auxiliary bladder type fuel tanks in each wing in accordance with FAA STC SA3634SW, revision 4, dated 30 March 2010. This EASA STC is a revision of LBA STC (EMZ) 0335/973, dated 5 May 1993.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval,</td>
<td>- Drawing List No. S10600, Revision N, dated January 27, 2008 - Installation Instructions No. S10665, Revision D, dated July 18, 1998 - Aircraft Flight Manual Supplements for Cessna P206 A, D and E model landplanes (SN P206-0161 through P206-00647), for aircraft with standard range bladder type fuel...</td>
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<td>KEYSTONE HELICOPTER CORPORATION</td>
<td>FAA STC SR2057NY-D</td>
<td>S90A</td>
<td>Installation of a tail mounted, forward facing camera.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval,</td>
<td>Installation Drawing List LC01702 rev. dated 1 March 2010 - Installation Instructions for Continuation/Extension/PM 01702 rev. dated 21 May 2010 - Flight Manual Supplement</td>
<td>None</td>
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<td>BUCHER LEICHTBAU AG (EASA.A.03797)</td>
<td>SEE EASA APPROVED MODEL LIST</td>
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<td>189 New Generation Stretch P/N 1120000-300 (NGS)</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC REF ST0373AT-D</td>
<td>GIV-X AND GV-SP</td>
<td>Installation of a Honeywell/EMS HD-710 combined SATCOM Swift Broadband High Speed Data System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval,</td>
<td>Install in accordance with Gulfstream Aerospace Index List noDICC110280009, Revision B, dated 10/06/09 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004004CF (or subsequent revisions or this decision)</td>
<td>Not Applicable</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10030691</td>
<td>(EASA.IM.A.S.03225)</td>
<td>FAA STC REF ST03685AT-D</td>
<td>G1V-SP (G550)</td>
<td>Installation of Flight Crew Sleeping Facilities.</td>
<td>JAR 25The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Install in accordance with Gulfstream Aerospace Index List no/GC5151/10000. Revision E, dated 06/18/09 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>Not Applicable</td>
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<td>ANORA INTERNATIONAL LLC</td>
<td>10030694</td>
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<td>FAA STC REF ST03646AT-A</td>
<td>B737-300</td>
<td>Installation of main deck cargo loading system.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Anora International LLC Control Data List DL84990 Rev. Cited June 2, 2010 Instructions for Continued Airworthiness. ICA Report 9444 Rev. For later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>None</td>
<td>01/07/2010</td>
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<td>BRITISH AIRWAYS</td>
<td>10030714</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>B737-400</td>
<td>Fuselage skin GPS antenna structural provisions</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: EASA.IM.A.S.03225.</td>
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<td>ZODIAC AEROSAFETY SYSTEMS</td>
<td>10030757</td>
<td>(EASA.IM.A.S.03236)</td>
<td>REV. 5</td>
<td>TCCA STC SAD0-23</td>
<td>DHC-8-100/200-300 SERIES</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Configuration 1: Installation of Aerazur Type 51 De-Icer Boots in accordance with the list of documents approved by EASA. Installation of Aerazur Type 51 De-Icer Boots in accordance with the list of documents approved by EASA. Installation of Aerazur Type 51 De-Icer Boots in accordance with the list of documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/12/2016</td>
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<td>BUCHER LEICHTBAU AG</td>
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<td>(EASA.IM.A.S.03712)</td>
<td>REV. 1</td>
<td>TABLE E043-01-17 SEE AIRCRAFT ALLOCATION</td>
<td>Change to STC EASA A.S.03712 through 1. Extension of the STC approval to all aircraft models of the EmbraerERJ 190-Serie. Installation approvals for specific MSNs of the EmbraerERJ 190 and 190-Serie. Standardization of the product documentation package of the 9G NGS XLSApX Stretchet 11200500- Ser...</td>
<td>CS-25 Amendment 20The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. -Master Document List E043-05-08 Rev B dtd. 22.02.2016-Aircraft Allocation Table E043-01-17 Rev NC dtd. 31.01.2016-User Manual 25-67-43 Rev. 02 dtd. Jan 26/2010 for later revisions of the above listed documents approved by EASA. -Par User Manual E017-01-08- This STC constitutes an approval for installation only for aircraft serial numbers as stated in the Aircraft Allocation Table. For additional aircraft serial numbers the requirements for full compliance to have been demonstrated through a design approval-installation of this...</td>
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<td>26/02/2018</td>
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<td>TCCA STC SH10-16</td>
<td>AS350D1 AS350B1, AS350B2, AS350B3, AS350BA, AS350D2</td>
<td>Installation of Latitude Technologies S100 Tracking System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. 1117-MDL-001 revision N C dated 7 April 2010 - Master Drawing List 1117-400/004 revision 01 dated 7 April 2010 - Instructions for Continued Airworthiness</td>
<td>None</td>
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<td>FAA STD: SR200356</td>
<td>Cargo Hooks Kit model 200-352-00</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>155-145-00 revision 1 dated 28 May 2010 - Master Drawing List; 120-140-00 revision 0 dated 9 September 2009 - Owners Manual; 123-035-00 revision 0 dated 18 November 2009 - Service Manual 121-054-00 revision 1 dated 11 June 2010 - Rotorkraft Flight Manual Supplement.</td>
<td>None</td>
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<td>BOMBARDEIR INC.</td>
<td>Installation, Interior Completions In-System Models</td>
<td>Transport Canada SoC reference C-11-0055 dated 17 February 2011. EASA STC No. 1003781.</td>
<td>None</td>
<td>Restricted to ASN 9348 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS-2. The INL system is compatible only to IEEE 802.11(b).</td>
<td>25/02/2011</td>
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<td>FOOCA</td>
<td>Installation of ARA-EC135-AC60 Ambulance and Rescue System Modifications under EASA format at revision 0.</td>
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<td>None</td>
<td>29/11/2011</td>
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<td>ROCKWELL COLLINS INC.</td>
<td>Installation of Rockwell Collins enhanced Total Entertainment System in accordance with System List 1309713-002/-004.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
<td>None</td>
<td>08/07/2010</td>
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<td>DYNNAVATION</td>
<td>MOD06F0001 050210 - New flaps and control system</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
<td>None</td>
<td>08/07/2010</td>
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<td>AEROLITE AG</td>
<td>EMS Kit AV1000SP REGA P/N 1090150-501 Changes related to Rev. 1. Introduction of the Prospa MD Retainer Instr. P/N 1090170-563 and an additional &quot;Training Configuration 2&quot; into the already certified EMSEquipment P/N 1090150-501, Modifications of BaggageCompartment Instr. P/N 1090160-...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<td>None</td>
<td>24/09/2013</td>
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### List of EASA Supplemental Type Certificates

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<td>FOOCA</td>
<td>Installation of ARA-EC135-AC60 Ambulance and Rescue System Modifications under EASA format at revision 0.</td>
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<tr>
<td>ROCKWELL COLLINS INC.</td>
<td>Installation of Rockwell Collins enhanced Total Entertainment System in accordance with System List 1309713-002/-004.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<td>DYNNAVATION</td>
<td>MOD06F0001 050210 - New flaps and control system</td>
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<td>AEROLITE AG</td>
<td>EMS Kit AV1000SP REGA P/N 1090150-501 Changes related to Rev. 1. Introduction of the Prospa MD Retainer Instr. P/N 1090170-563 and an additional &quot;Training Configuration 2&quot; into the already certified EMSEquipment P/N 1090150-501, Modifications of BaggageCompartment Instr. P/N 1090160-...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<td>10030947</td>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td></td>
<td>206H, T206H</td>
<td>Provisioning for implementation and operation with a &quot;Turtle Pac160-ferry tank&quot;</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to the STC.</td>
<td>EASA STC No. 1704028030-01A, dated 28.06.2010</td>
<td>Installation of the ferry tank is not subject of this STC.</td>
<td>14/07/2010</td>
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<td>10030918</td>
<td>GAMA ENGINEERING LTD.</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>EXTRA EA-300 / L</td>
<td>External camera installation</td>
<td>The Certification Basis for the original product remains applicable to the STC.</td>
<td>Master Documents Drg Mdl. 1054-01 Issue 1, TCCA Issue Paper IP09075-1, Revision A, dated 25.03.2010 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
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<td>10030900</td>
<td>BOMBARDIER INC.</td>
<td>REV. 3</td>
<td>TCCA STC SA05-60</td>
<td>BD-700-1A11</td>
<td>Installation, Interior Completions - Service Improvements Modifications - ASN 9359</td>
<td>EASA TCDS No. IM.A.009 The Certification Basis for the original product remains applicable to the STC.</td>
<td>Restricted to ASN 9359</td>
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<td>SKYTRAC SYSTEMS LTD.</td>
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<td>TCCA STC SA09-93</td>
<td>DHC-6-300/DHC-6-200/DHC-6-100</td>
<td>SkyTrac Flight Following / SATCOM Installation in acc. with TCCA STCSA09-80 ISAT-100 transceiver installed in aft zone of aircraft, Dual element GPS/Intrum antenna installed on upper fuselage, optional DIVI-250 and CDP-250 control panels installed at centre radio panel.</td>
<td>The Certification Basis for the original product remains applicable to the STC.</td>
<td>Master Drawing List MLD09074, Revision A, dated May 12, 2010</td>
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<td>SKYTRAC SYSTEMS LTD.</td>
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<td>TCCA STC SA09-80</td>
<td>CESSNA 208, 208B</td>
<td>SkyTrac Flight Following / SATCOM Installation in acc. with TCCA STCSA08-93 ISAT-100 transceiver installed in aft zone of aircraft, Dual element GPS/Intrum antenna installed on upper fuselage, optional DIVI-250 and CDP-250 control panels installed at centre radio panel.</td>
<td>The Certification Basis for the original product remains applicable to the STC.</td>
<td>Master Drawing List MLD09075, Revision A, dated May 12, 2010</td>
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<td>10030934</td>
<td>BOMBARDIER AEROSPACE</td>
<td>REV 1</td>
<td>TCCA STC SA10-50</td>
<td>BD100-1A10</td>
<td>Activation of Wireless Local Area Network Access Point on Bombardier BD-100-1A10 Revision of the EASA STC 10030934, EASA Validation of TC1A approved Supplemental Type Certificate SA10-50</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
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<td>10030938</td>
<td>BOMBARDIER AEROSPACE</td>
<td>REV 1</td>
<td>TCCA STC SA10-44</td>
<td>BD100-1A10 CHALLENGER 300</td>
<td>Installation of Mechanical Provisions and LRU Installation AirNet4Go-to-Ground Communication System and Immersat SAttom System Revision of the EASA STC 10030938 [&lt;sup&gt;1&lt;/sup&gt;] validation of TCA approved Supplemental Type Certificate SA10-44</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>None</td>
<td>23/09/2010</td>
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<td>10030938</td>
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<td>10030942</td>
<td>LIFEPORT, LLC</td>
<td>FAA STC ST01891LA-D</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Various Medical Stretchers/Validation of FAA STC ST01891LA-D</td>
<td>Multi-Models Commercial Medical Stretcher/Validation of FAA STC ST01891LA-D</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. Certification basis as per TCD 8-1 [&lt;sup&gt;2&lt;/sup&gt;] addressed per TGM 29/12/CRI H-01 Special Condition H-01 &quot;Enhanced Avionics...&quot;</td>
<td>FAA approved Model List (AML) No. ST01891LA-D, dated May 27, 2010 - FAA approved Airplane Flight Manual Supplement No. EGR-808-11, Rev Dated May 27, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 200404/CF (or subsequent revisions off this decision).</td>
<td>15/07/2010</td>
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<td>10030951</td>
<td>TASS Inc.</td>
<td>FAA STC ST00593SE-D</td>
<td>B747-400</td>
<td>Active Noise Reduction Headset</td>
<td>Installation Provisions in accordance with BF Goodrich Master Drawing List D2000-396, sheet 1, Revision New, dated November 22, 2000 or later FAA/FAASGS approved validation. Validation of FAA STC ST00593SE-D</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>None</td>
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<td>10030954</td>
<td>ICELANDAIR TECHNICAL SERVICES</td>
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<td>GAS AIRVAN</td>
<td>Installation of TopScan Camera System</td>
<td>CS-23, Amdt. 1, is elected to comply with for the items affected by the change. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval</td>
<td></td>
<td></td>
<td>None</td>
<td>Only applicable for aircraft/ preconditions as defined in Engineering Order, Doc. No. EO-GTA-2520-03, Rev. IR, dated 28-Jun-2010; or later approved revision.</td>
<td>15/07/2010</td>
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<td>STC Holder</td>
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<td>WESTERN AVIONICS</td>
<td>TCJA STC-SA04-96</td>
<td>DHC-8-801-202 DHC-8-201-202 DHC-8-102-103-106 Second ADF installation TCJA STC SA04-96, Issue 1 dated 23 December 2004 FAR 25 as per EASA M.A. 191 The Certification Basis for the original product remains applicable to theses certificate/app. Approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product an unchanged and remain applicable to this ce...</td>
<td>None</td>
<td>16/07/2010</td>
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<td>SAFRAN CABIN BELLINGHAM, INC.</td>
<td>FAA STC ST03875E</td>
<td>B747-400 LCD Monitor installation on Boeing Model 747-400 series aircraft inacc...</td>
<td>ICA Supplement HTM57 Revision A, dated March 1, 2011 or later revisions of the above listed documents approved by EASA inaccordance with EASA Commission Decision 2004/649/EC (or subsequentrevisions of this decision)</td>
<td>None</td>
<td>16/07/2010</td>
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<td>10030956</td>
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<td>AVIONICS INTEGRATION &amp; ENGINEERING CO R I P O R A T I O N AG (AIEC)</td>
<td>BEECH 200</td>
<td>Installation of HySpex Camera System (NEO)</td>
<td>-EASA approved Master Data List, Doc. No. CV877-049, Rev A, dated 14-Jul-2010; or later approved version-EEASA approved Engineering Order, Doc. no. CV977-062, Rev A, dated 14-Jul-2010; or later approved version-EEASA approved Airplane Flight Manual Supplement, Doc. No. CV977-071, Rev. R or later ap...</td>
<td>Only airtaxis in compliance with the preconditions as listed inChapter C of Engineering Order. Doc. No. CV977-062, Revision A, dated 14-Jul-2010, or later approved revision, are eligible for this change.</td>
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<td>JET AVIATION AG</td>
<td>REV. 2</td>
<td>A319-115 Airbus A319-115 &quot;BQJ03632&quot; VIP Cabin Interior Completion/Revision 1: Update of the Instructions for Continued Airworthiness/Revision 2: Bedroom Reconfiguration Transformation of bedroom fromNon-TTOL to TTOL area by Reconfiguration Transformation of Airworthiness/Revision 2: Bedroom Completion. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the product are unchanged and remain applicable to this certificate...</td>
<td>-JUBL-M-01444DLC-003 ISSUE B MASTER DATA LIST BsicJ03632-01-015B UE.C AIRPLANE FLIGHT MANUAL SUPPLEMENTBCJ03632-1 55-01 ISSUE C INSTRUCTIONS FOR CONTINUED AIRWORTHINESS/JUBL-M-01444CCS-003 ISSUE A CERTIFICATION COMPLIANCE SHEET or later revisions of the above listed document(s) approved/accepted under the...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherpreviously installed change and/or repair willnot introduce no adverseeffect upon the airworthiness of the product.</td>
<td>11/05/2023</td>
<td>Active</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 1</td>
<td>CESSNA U 206G, TU 206G Installation of Propeller MTV-5-1-E-C-F-R (80)CPR20-54 ohMTV-5-1-E-C-F-B/CF220-54 MTV-5-1-E-C-F-R(My CPR20-54-Instructions for Continued Airworthiness and Installation Instructions)E-1031-Airplane Flight Manual Supplement N...</td>
<td>E-1032MTV-5-1-E-C-F-B/CF220-54-Instructions for Continued Airworthiness and Installation Instructions)E-1116-Airplane Flight Manual Supplement N...</td>
<td>1. This STC is applicable to Cessna U 206G, TU 206G both equipped withR/R Allison 250- C205 (Solsy Turbine PAC) in accordance with LSA EM2 (STC) 0082/073a only 2. This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraph &quot;Description&quot; Com...</td>
<td>19/07/2010</td>
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<td>SAFRAN CABIN BELLINGHAM, INC.</td>
<td>FAA STC ST02066SE</td>
<td>B747-400</td>
<td>Seat Installation and interior reconfiguration on Boeing Model 747-400 series aircraft in accordance with Health Techos, Inc. Master Drawing L174PD-CL-173 Revision B dated May 11, 2010, or later FAA approved revision. Validation of FAA STC ST02066SE.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: FAR 25.585(a)(11) - FAR 25.833(b)(1) - FAR 25.889(a)(11) - FAR 25.593(b)(1) - FAR 25.1353(a)(11) - FAR 25.1353(b)(11).</td>
<td>ICA Supplement HTM366 Revision E, dated May 20, 2010.</td>
<td>None</td>
<td>19/07/2010</td>
<td>Active</td>
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<tr>
<td>RUAG AG</td>
<td>REV. 1</td>
<td>P. 180 AVANTI II</td>
<td>In-Flight Entert. System installation by installing a ROSEN/Rosier/Viewix moving Map and DVD System, an IR-Eye, Inc FLIGHT DISPLAY/STEM 10.2 inch LCD and two 7 inch LCD’s and a FLIGHT DISPLAY SYSTEM wireless audio transmitter for audio distribution to SENNHEISER HDR140 wireless Headsets.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List: Nr. 163206-00, Rev. 000, 18.07.10-Instructions for Continuous Airworthiness: Nr. 163206-00, Rev. 000.02.10.09-Declaration of Compliance: Nr. 163415-410, Rev. 000.</td>
<td>None</td>
<td>02/02/2021</td>
<td>Active</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>L-410 UVP- E, L-410 UVP-E, L-410 UVP-E20 CARGO</td>
<td>Introduction of under wing emergency exits.</td>
<td>CS-23 Amendment 1 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Compliance Record: S21.COMP-0704 Issue 1 Dated 15 July 2010/Manual Supplement: S21.MOL-0704 Dated 5 July 2010/Flight Manual Supplement: S21.FMS-002 Issue 1 Dated 15 July 2010/Supplement Manual: S21.TEC-023 Issue 1 Dated 15 July 2010 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>11/06/2012</td>
<td>Active</td>
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<tr>
<td>AEROLITTE AG</td>
<td>REV. 3</td>
<td>MBB-BK 117 C-2</td>
<td>EMS Equipment P/N: 143541-501-502.This Major Change to EASA STC 10030982. Rev. 2 introduces the following modifications: Components to EMS Equipment P/N: 143541-501-502. For P/N 143541-501-502.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements is applicable to this certificate/approval: Equivalent Safety Finding(s) CRI D-1 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>For configuration P/N: 143541-501 Flight Manual Supplement FMS-10-003 Rev. 0/Operation and Installation Manual OPM-10-04 Rev. 0/For configuration P/N: 143541-502 Flight Manual Supplement FMS-11-001 Rev. 0/Operation and Installation Manual OPM-11-01 Rev. 0/Bor later revisions of the above listed documents approved by EASA.</td>
<td>Compatibility with NVIS has not been demonstrated. Compatibility with TC holder. 3-seat bench has not been demonstrated.</td>
<td>26/01/2017</td>
<td>Active</td>
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<tr>
<td>AMSAFE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC SA02276AK</td>
<td>Installation of AMSAFE Inflatable Restraint Systems in accordance with Model Specific Master Data List identified on AMSA02276AK.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- AMSAFE Flight Manual Supplement Document Number ES00433. Revision NG, dated 25 August 2006/ (&lt;, &gt;)/AMSASF Instructional Continuous Airworthiness Document Number ES00424. Revision F, dated 3 June 2019/ (&gt;)/or later revisions of the above listed documents aproved/approved onwards of EASA in acc.:</td>
<td>1). The STC is valid for those aircraft types and models approved by EASA as defined on EASA Product Lists, that are contained in the FAA Approved Model List SA02276AK, rev. T, dated 30 December 2016 or later FAA Approved revision 2). If the holder agrees to permit another person to use this certificate/approval.</td>
<td>13/02/2018</td>
<td>Active</td>
<td>10031010</td>
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<td>ZIM FLUGSITZ GmbH</td>
<td>REV. 1</td>
<td>FAA STC SA02276AK</td>
<td>Replacement of E/C seats by ZIM E/C seats. Type ECO-01/This STC consists in reusing the existing seats by Type ECO-01seat.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Declaration of Compliance, ref: 50201-COC-003 Issue GA, dated 27 July 2010 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>21/07/2010</td>
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### List of EASA Supplemental Type Certificates

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<th>STC ref</th>
<th>Model</th>
<th>Description</th>
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<th>Limitations/Conditions</th>
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<td>TELEDYNE CONTROLS, LLC</td>
<td>REV. 1</td>
<td>FAA STC ST04988A</td>
<td>B767-200</td>
<td>Installation of Aircraft Communications, Addressing and Reporting System (ACARS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval Teledyne Controls’ Master AFM Supplement 65-A90, B90 and C90 airplanes with PT6A-28 engines</td>
<td>23/07/2010</td>
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<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA SA47055W</td>
<td>65-A90, B90, C90</td>
<td>Upgrade King Air 65-A90, B90 and C90 airplanes with PT6A-28 engines</td>
<td>FAR 23 Installation of Pratt and Whitney PT6A-27 or PT6A-28 engines in Model C90 in accordance with drawing 02-001, Revision B dated June 5, 2008 or later approved revisions, or PT6A-27 or PT6A-28 engines in Model 65-A90 or B90 in accordance with drawing 04-001, Revision B dated June 5, 2008 or later approved revisions</td>
<td>Engines can be either type certified Pratt and Whitney PT6A-27 or PT6A-28 engines, or converted PT6A-27 or PT6A-28 engines from PT6A-28 engines in accordance with the abovementioned documents approved by EASA</td>
<td>26/07/2010</td>
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<td>BOMBARDIER INC.</td>
<td>REV. 1</td>
<td>TCCA STC SA05-60</td>
<td>BD-700-1A11</td>
<td>Installation, Interior Completions- In Service Improvements Modifications-ASN 9373</td>
<td>EASA TCDS No. IM.A.009The Certification Basis for the original product remains applicable to this certificate/ approval The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval</td>
<td>Restricted to ASN 9373</td>
<td>24/02/2011</td>
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<td>DANSK FLY ELEKTRONIK ApS</td>
<td>PA-42, PA-42-720, PA-42-1000</td>
<td>Installation of Avadyne EX500 MFD</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval DFE Design Order DFE-DO-0977 Design Package Rev 00 or later revisions</td>
<td>DFE Design Order DFE-DO-0977 Design Package Rev 00 or later revisions</td>
<td>None</td>
<td>27/07/2010</td>
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<td>DANSK FLY ELEKTRONIK ApS</td>
<td>PA-42, PA-42-720, PA-42-1000</td>
<td>Installation of one or two Garmin GTX 330D Mode-S Transponder</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval DFE Design Order DFE-DO-0977 Design Package Rev 00 or later revisions</td>
<td>DFE Design Order DFE-DO-0977 Design Package Rev 00 or later revisions</td>
<td>None</td>
<td>27/07/2010</td>
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List of EASA Supplemental Type Certificates

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<td>BOMBARDIER INC.</td>
<td>REV 1</td>
<td>TCSTA STC SA08-65</td>
<td>BD700-1A10</td>
<td>Water Tank Modification</td>
<td>EASA TCSD No.</td>
<td>EASA.IM.A.S.03036 dated 02 June 2009 ; TCSTA STC No. SA08-65 Issue 3 dated 02 July 2010 ; TCSTA STC No. SA08-65 Issue 3 dated 21 July 2010 ; Aero Consulting Services Modification Data Summary Number F25000/Revision C or later approved revision; Bombardier Instructions for Continued Airworthiness.</td>
<td>None</td>
<td>21/07/2010</td>
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<td>AGUSTAWESTLIA CORPORATION</td>
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<td>WIPAIRE Inc.</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
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28/07/2010
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10031129

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<tr>
<td>ACS-NAI Ltd</td>
<td>10031149</td>
<td>TCCA STC-SA10-26</td>
<td>CL-600-2B10 Installation, CMC Electronics Pilot view Class II Electronics Flight Bag EASA M.A.0237 for the original product remains applicable to this certificate/approval. If required, a change in speeds for the electrical system per EASA.IM.A.023 requires a separate STC.</td>
<td>None</td>
<td>29/07/2010</td>
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<td>AVIONIK STRAUBING ENTWICKLINGS GmbH</td>
<td>10031180</td>
<td>CS-23</td>
<td>2. PA-31T2 Dual/single installation of Garmin GN9350A(KS30/30/40) with GPS converter DAC GDC31.</td>
<td>none</td>
<td>13/06/2010</td>
<td>Active</td>
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<tr>
<td>F. LIST GERMANY GmbH</td>
<td>10031182</td>
<td>CS-23</td>
<td>2. PA-31T2 Dual/single installation of Garmin GN9350A(KS30/30/40) with GPS converter DAC GDC31.</td>
<td>none</td>
<td>08/12/2017</td>
<td>Active</td>
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<td>BOMBARDIER INC.</td>
<td>10031184</td>
<td>CS-23</td>
<td>BD-700-1A10 Installation, Complete Custom Aircraft Interior - ASN 9551 Transport Canada SoC reference C-10-0555 dated 29 July 2010.TCCA STC No. SA04-113 Issue 1 dated 28 October 2004.Aero Consulting Services Configuration Definition List D964000 Revision P dated 27 July 2010 but restrictions for Continued Airworthiness GC93510000/0/42 remain applicable to this certificate/approval.</td>
<td>Restricted to ASN 9351 and Refer to the Limitations sections of the required AFM Supplement and the referenced FCOM/2.WLAN system is compatible only to IEEE 802.11a/b/g. Prior to installation of this modification the installer must determine that the interrelationship between this modification and...</td>
<td>30/07/2010</td>
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List of EASA Supplemental Type Certificates

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<tr>
<th>STC No.</th>
<th>STC Holder</th>
<th>STC Ref.</th>
<th>STC serial No.</th>
<th>Type</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tbody>
<tr>
<td>10031201</td>
<td>STC Holder</td>
<td>A340-213</td>
<td>A340-200</td>
<td>-</td>
<td>Installation of provisions for Live TV, Satcom, and W-AN ihrer this STC installs several structural provisions for Live TV, Satcom, and W-LAN systems, relocates the existing upper Anti-Collision Light (ACL) and installs a additional upper ACL.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>FAA STC 04-1070M, Issue 1 dated 22 July 2011, Installation for Continued Airworthiness, ref: A-00-10_1311-ICA</td>
<td>Limited to MSN 151</td>
<td>30/07/2010</td>
<td>Active</td>
<td>10031201</td>
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<tr>
<td>10031209</td>
<td>EASA.A.S.03618</td>
<td>REV. 4</td>
<td>382</td>
<td></td>
<td>Installation of Saab Missile Protection System. The installation consists of a forward sensor fairing on both sides of the nose and a rear sensor installation in the tail. The dispensers are installed under both sides under the rear edge of the wing. Inside the aircraft a control panel has been installed.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: CS 25-65, CS 25-66, CS 25-67. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Serial 4954 Only</td>
<td>12/06/2023</td>
<td>Active</td>
<td>10031209</td>
<td></td>
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<tr>
<td>10031215</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A340-300</td>
<td>Structural Installation of 2 IFF Antennas</td>
<td>CS 25 Amendment 6</td>
<td>The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certificate of Design MA 1903, FAA STC 09040SC dated 02/05/2010</td>
<td>none</td>
<td>none</td>
<td>02/06/2010</td>
<td>Active</td>
<td>10031215</td>
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<tr>
<td>10031221</td>
<td>FAA STC 09040SC</td>
<td>B737-300</td>
<td>Structural Installation of a digital Flight Data Recorder Unit (DFDR)</td>
<td>CS 25 Amendment 6</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>FAA STC 09040SC dated 09/08/2006</td>
<td>FAA STC 09040SC dated 09/08/2006</td>
<td>none</td>
<td>03/08/2010</td>
<td>Active</td>
<td>10031221</td>
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<tr>
<td>10031227</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A340-300</td>
<td>Structural Installation of a VIHF Antenna</td>
<td>CS 25 Amendment 6</td>
<td>The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certificate of Design MA 1904, FAA STC 09040SC dated 09/08/2006</td>
<td>Certificate of Design MA 1904, FAA STC 09040SC dated 09/08/2006</td>
<td>none</td>
<td>03/08/2010</td>
<td>Active</td>
<td>10031227</td>
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<tr>
<td>10031230</td>
<td>FAE</td>
<td>SA12TQL</td>
<td>Installation of wing root fairings in accordance with FAA STC 09040SC</td>
<td>CS 25 Amendment 6</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Knuts2U Ltd. Installation and Maintenance Manual, documentation number 020IRM-M, revision E, dated May 2, 2010</td>
<td>none</td>
<td>FAA STC 09040SC dated 09/08/2006</td>
<td>03/08/2010</td>
<td>Active</td>
<td>10031230</td>
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**List of EASA Supplemental Type Certificates**

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<tr>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC no.</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td><strong>10031235</strong></td>
<td></td>
<td>FAA STC SA1521QL</td>
<td>2. PA-32R-301, PA-32R-301T</td>
<td>Installation of flaps, aileron, flap/aileron and stabilator gap seal in accordance with FAA STC SA1521QL.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>None</td>
<td>03/08/2010</td>
<td>Active</td>
<td>10031235</td>
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<tr>
<td><strong>10031236</strong></td>
<td></td>
<td>FAA STC SA195QL</td>
<td>2. PA-32RT-300T</td>
<td>Installation of Flap Hinge Fairing seals in accordance with FAA STC SA195QL.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>None</td>
<td>03/08/2010</td>
<td>Active</td>
<td>10031236</td>
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<tr>
<td><strong>10031240</strong></td>
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<td>FAA STC SA3226NM</td>
<td>2. PA-32RT-300T</td>
<td>Installation of 16.25 US Gallon usable fuel lightning resistant wing tip tanks</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>-Master Data List MCL-0080-16 Rev. 00 dated 23/5/2010 (plus EDX10105822)-Service Bulletin BA-737-34-015 Rev. 00-6Fmanual change request, BA Ref: 100988651</td>
<td>05/08/2010</td>
<td>Active</td>
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<tr>
<td><strong>10031241</strong></td>
<td></td>
<td>FAA STC SA1521QL</td>
<td>2. PA-32R-301, PA-32R-301T</td>
<td>Installation of cockpit sunvisors</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>None</td>
<td>03/08/2010</td>
<td>Active</td>
<td>10031241</td>
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<tr>
<td><strong>10031247</strong></td>
<td></td>
<td>FAA STC SA0871SE</td>
<td>SEE FAA AMD 15-2010</td>
<td>Installation of cockpit sunvisors</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>None</td>
<td>05/04/2023</td>
<td>Active</td>
<td>10031247</td>
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</tbody>
</table>

**EASA STC no.**

- **KNOTS ACQUISITION LTD.**
- **FLINT AEROLTD.**
- **SKYTRAC SYSTEMS LTD.**
- **ROSEN SNUISOR SYSTEMS LLC.**

**EASA STC no.**

- **EASA IM A.S. 02712**
- **EASA IM A.S. 03270**
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<tr>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC no.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<tbody>
<tr>
<td>ARMITIVE ENGINEERING CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SE101256C</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of missile, valve springs, ni-resist valve guide, exhaust valve with chrome flash on stem, and motor shaft with thrust buttons in accordance with EG drawings AEC63114, Rev E, dated February 6, 2003, or later FAA approved revision - EASA Validation of FAA STC/SE101256C</td>
<td>CAR 13 / FAP 337 The Certification Basis for the original products remain applicable to this certificate.</td>
<td>none</td>
<td>none</td>
<td>04/11/2010</td>
<td>Active</td>
<td>10031253</td>
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<tr>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>DC-9-83 (MD 83)</td>
<td>Cabin Layout/Interior Change - DC-9-83</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>none</td>
<td>none</td>
<td>05/06/2010</td>
<td>Active</td>
<td>10031258</td>
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<tr>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td>C90A, C90GT, C90GTI, C90</td>
<td>Installation of an Axiline Traffic Advisory System TAT 600/106/20</td>
<td>The Certification Basis of the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Master Document List, Project No. C90-NY-07137, Rev. 1, dated 3 August 2010, or later approved revision</td>
<td>none</td>
<td>Prior to installation of this design change it must be determined that the interpersonal relationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10031264</td>
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<tr>
<td>LEARJET Inc.</td>
<td>FAA STC SA117NM-D</td>
<td>CL-600-2B16</td>
<td>Installation of the RH Forward Lavatory</td>
<td>EASA MA 0237 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/ or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>FAA STC No. SA117NM-D dated 24 June 1998, Learjet Master Drawing List No. CA-3068a-TUC. Revision A dated 10 June 1998 or later approved revision, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2000/44/CF (or subsequent revisions of this decision)</td>
<td>none</td>
<td>none</td>
<td>06/08/2010</td>
<td>Active</td>
<td>10031271</td>
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<tr>
<td>AGUSTAWEStLA ND PHILADELPHICO RPORATION</td>
<td>FAA STC</td>
<td>SR02265NY</td>
<td>A119/A119MKI</td>
<td>Installation of pedal extensions on A119, A119MKI helicopters</td>
<td>none</td>
<td>none</td>
<td>06/06/2010</td>
<td>Active</td>
<td>10031274</td>
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<tr>
<td>RAIBECk ENGINEERING Inc.</td>
<td>REV. 5</td>
<td>FAA STC SA268NM-S</td>
<td>(EASA.12.A.00766)</td>
<td>Installation of a set of Hartbell/Raisbeck Quick Turfbon Propellers onto swept Turbofan Propellers or Five-Blade Swept Turbofan Propellers. The installation consists of Hartbell/Raisbeck 4-bladed or 5-bladed propellers and aluminum hubs with spinners, and a propeller drive systems. Revision to STC 1003.</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change with an impact on the noise database and/ or emissions levels. The Environmental protection requirements are amended as follows: A30 Annex 16A.</td>
<td>Installation in accordance with Raisbeck Engineering Drawing List 85-1500, Revision BV, dated September 13, 2019, Operation in accordance with the applicable FAA approved Airplane Flight Manual Supplement (AFMS) in accordance with AFM, SA2688NM-S, or previsions of this STC. Continued Airworthiness/None.</td>
<td>14/11/2023</td>
<td>Active</td>
<td>10031275</td>
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<td>EASA STC no.</td>
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<td>10031277</td>
<td>AGUSTAWESTLA ND-PHILADELPHIACO RPORTATION</td>
<td>FAA STC SR02769NY</td>
<td>AW119MKII</td>
<td>Installation of a Border Guard Configuration</td>
<td>09/08/2010</td>
<td>Active</td>
<td>10031277</td>
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<td>10031298</td>
<td>AERO DYNAMIX, Inc.</td>
<td>FAA STC SR09519RC</td>
<td>AW119MKII</td>
<td>Installation of Night Image Vision System (NVIS)</td>
<td>03/07/2024</td>
<td>Active</td>
<td>10031298</td>
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<td>10031299</td>
<td>LEARJET Inc.</td>
<td>FAA STC SA8118NM-D</td>
<td>CL-600-2B16</td>
<td>Installation of the RH Forward Galley</td>
<td>06/08/2010</td>
<td>Active</td>
<td>10031299</td>
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<td>10031307</td>
<td>RAISBECK ENGINEERING Inc.</td>
<td>FAA STC ST01182LA</td>
<td>A320-232</td>
<td>A320 bin extension system installation of a bin extension system in accordance with FAA Supplemental Type Certificate ST01182LA issued 10 July 2001.</td>
<td>06/08/2010</td>
<td>Active</td>
<td>10031307</td>
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<td>EASA STC no.</td>
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<td>Associated Technical Documentation</td>
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<td>10031318</td>
<td>LEARJET INC</td>
<td>FAA STC</td>
<td>SAB119NM-D</td>
<td>CL-600-2B16</td>
<td>Installation of the BuhlHead F.S.353</td>
<td>EASA M.A.023 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. SAB119NM-D dated 24 June 1998.</td>
<td>none</td>
<td>10/08/2010</td>
<td>Active</td>
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<tr>
<td>10031319</td>
<td>LEARJET INC</td>
<td>FAA STC</td>
<td>SAB120NM-D</td>
<td>CL-600-2B16</td>
<td>Interior Installation - Main Cabin</td>
<td>EASA M.A.023 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. SAB120NM-D dated 25 June 1998.</td>
<td>none</td>
<td>10/08/2010</td>
<td>Active</td>
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<tr>
<td>10031320</td>
<td>LEARJET INC</td>
<td>FAA STC</td>
<td>SAB122NM-D</td>
<td>CL-600-2B16</td>
<td>Installation of the Armledge / Dado</td>
<td>EASA M.A.023 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. SAB122NM-D dated 24 June 1998.</td>
<td>none</td>
<td>10/08/2010</td>
<td>Active</td>
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<tr>
<td>10031322</td>
<td>RUAG AG</td>
<td>REV. 1</td>
<td>580 N</td>
<td></td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List: RUAG Doc. Nr. 186441-00-Instructions for Continued Airworthiness; RUAG Doc. Nr. 186441-00-Rotorcraft Flight Manual Supplement; RUAG Doc. Nr. 186441-00-Aircraft Configuration as defined in the above listed documents/approved acceptance/airworthiness/approved configuration for the product.</td>
<td>Serial number: 5M08</td>
<td>18/01/2021</td>
<td>Active</td>
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<tr>
<td>10031332</td>
<td>FOKKER TECHNIK B.V.</td>
<td>REV. 1</td>
<td>A318-112</td>
<td>A318-112 VIP conversion, for description see Chapter 1.1 of Compliance Record Report A18-MAC-CRR-0001 issue 5.d.d.</td>
<td>A318-112 VIP conversion, for description see Chapter 1.1 of Compliance Record Report A18-MAC-CRR-0001 issue 5.d.d.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Compliance Record Report A18-MAC-CRR-0001 issue 5.d.d.</td>
<td>1. Refer to Chapter 2, Limitations of the AFM Supplement Airbus A318 Model A318-112 Aircraft for the VIP Conversion As installed in Serial number 3238.</td>
<td>05/12/2019</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Foreign/NAA STC ref.</td>
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<tr>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 2</td>
<td>MYSTERE FALCON 20 ES, 20 GF, MF50, MF90, C550 CITATION II</td>
<td>Installation of a multi-functions display Honeywell type KMD-85 applicable to Cessna Citation II model 550 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/related noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>UAE 015 of 23 PA or later approved revisions.</td>
<td>Implemented in accordance with Document Number: Installation Instruction I-2023163, Rev.2, dated 17th June 2022.</td>
<td>03/01/2015</td>
<td>Active</td>
<td>10031332</td>
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<tr>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 3</td>
<td>208B</td>
<td>208B Ferry Fuel Tank Installation – Rev. 3/2r: Introduces a 300 Gal Ferry Tank and electrical connection of auxiliary fuel pump to existing 28 VDC auxiliary power outlet. Rev.3: Administrative corrections to STC limitations.</td>
<td>The Type and OS2 Certification Bases (CB) for the original product and improvements applicable to this certificate/approval. The requirements for environmental protection and the associated/related noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installed and maintained in accordance with Document Number: Flight Manual Supplement, SPOH, 2023163_3, Rev.2, dated 17th June 2022, or later revisions of the above listed document.</td>
<td>This STC is applicable to Textron Aviation Cessna 208B aircraft with TPE331-12JR engines installed. Pratt &amp; Whitney PT6A-114A or -140. Prior to installation of this STC, the original STC limitations.</td>
<td>23/02/2023</td>
<td>Active</td>
<td>10031342</td>
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<tr>
<td>E.A.D AEROSPACIE AIRWORTHINESS</td>
<td>REV. 1</td>
<td>A320</td>
<td>Installation of Advanced Mission Display System Phase A - AMDIS/A on A320 Series Aircraft. Revision of the EASA STC 100313540 to remove EWS5 limitation.</td>
<td>In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CPI H-01 Special Condition H-01 Enhanced Airworthiness Program for A320 family (ICAO on EWS5).</td>
<td>- Master Drawing List ref. 01272-EAS-A-AMDL, Revision C, dated 22nd November 2010.</td>
<td>None</td>
<td>29/11/2010</td>
<td>Active</td>
<td>10031354</td>
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<tr>
<td>ROCKWELL COLLINS INC.</td>
<td>FAA STC ST01526A</td>
<td>B747-400</td>
<td>Installation of Rockwell Collins enhanced Total Entertainment System in accordance with System List 130461H. Revision N/C, dated August 16, 2005 or later.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Instructions for a Continued Airworthiness document rev. 1304494 Rev. 2, Aircraft Maintenance Manual Supplement 4/2012, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/45/C (or subsequent revisions of this decision).</td>
<td>None</td>
<td>11/08/2010</td>
<td>Active</td>
<td>10031359</td>
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<tr>
<td>VAN RIEMSDUIK ROTTERDAM B.V.</td>
<td>REV. 9</td>
<td>B747-200F/400F-400BF F-8F 747-400, 747-8F 747-200F, 747-400F</td>
<td>RZx-7186 Special Cargo Container 125’x196’ Approval of NOE-0506 Special Cargo container 125’x196’ with Part Number 136.020 and RZX-0506 Special Cargo container 125’x196’ with Part Number 136.020 for use in B747/8 freighter. Rev.5: Minor changes improving fuel level indicator and generator fuel bin sys...</td>
<td>The Type and OS2 Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated/related noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval w...</td>
<td>Compliance Record Ref. CR-0013 magical document Statement CST 0506 issue 2 - CMM / IPL RZX Container P/N 136.020-000, Ref. 25-55-06 / RMS06-85/00a-5 - CMM / IPL RZX Container P/N 136.020-000 and P/N 136.050-000, Ref. 25-55-06 / RMS06 issue 2b. CMM / IPL RZX Container P/N 136.050-001, Ref. 25-55-0...</td>
<td>Applicability to Model 747-400 limited to cargo configuration as per Boeing Service Bulletin 747-00-2004 &quot;747-400BCP&quot; or 747-400SF. Prior to installation of this change, it must be determined whether the F/E/3/T relationship between the change/repair and any other previously installed change/repair is applicable.</td>
<td>19/06/2024</td>
<td>Active</td>
<td>10031360</td>
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<tr>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACS5)</td>
<td>FAA STC ST01917A</td>
<td>B747-400</td>
<td>Installation of ACSS Mode S Transponder with ELT/ES/MS capability (Mode S Transponder in existing ACSS Traffic Alert and Collision Avoidance System). TCA 3. STC in accordance with FAA approval ACSS Master Data List Document No. 8900019-001, Revision B, dated August 2, 2006 of Index of Master Data List...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
<td>See Limitation Section of FAA approved ACSS Master Data List No. 8900019</td>
<td>11/08/2010</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>10031371</td>
<td>Coupling of the existing Sagem/SFIM Autopilot installation with the Garmin GNS530</td>
<td>The Certification Basis for the original product remains applicable to this certificate</td>
<td>none</td>
<td>12/02/2010</td>
<td>Active</td>
<td>10031371</td>
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<td>DANSK FLY ELEKTRONIK ApS</td>
<td>10031404</td>
<td>Installation of L-3 Skywatch HP Traffic Alert &amp; Collision Avoidance System, Model SKY999A</td>
<td>The Certification Basis for the original product and the associated/certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate</td>
<td></td>
<td>13/08/2010</td>
<td>Active</td>
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<tr>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD</td>
<td>10031405</td>
<td>ICU System Installation with ViaSat VMT 1500 Ku Band System and Honeywell MHC Cabin Management</td>
<td>EASA MA.009 The Certification Basis for the original product remains applicable to this certificate and approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate.</td>
<td>Restricted to ASN 9287</td>
<td>23/05/2011</td>
<td>Active</td>
<td>10031405</td>
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<tr>
<td>RAM AIRCRAFT, LP</td>
<td>10031426</td>
<td>Installation of vortex generators in acc. with FAA STC SA125SW-D, last issue, dated May 12, 2008 (Option 1) Modification in acc. with RAM Master Drawing No. 1507, Revision C, dated 11/03/1992 (Option 2) Modification in acc. with RAM Master Drawing No. 1507, Revision C, dated 11/03/1992(a) not approved)</td>
<td>The Certification Basis for the original product remains applicable to this certificate and approval. This certificate is approved with the following environmental protection and/or a change to the associated/certified noise and/or emissions levels.</td>
<td>None</td>
<td>12/11/2010</td>
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<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>10031457</td>
<td>First and Business Class Cabin Interior - BA 777-300 aircraft</td>
<td>EASA TDCS No. MA.003 The Certification Basis for the original product remains applicable to this certificate and approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate</td>
<td>None</td>
<td>17/08/2010</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>10031459</td>
<td>Installation of Thales I-MASTER A-Flight Radar Turret</td>
<td>EASA TDCS No EASA.A.1097 The Certification Basis for the original product remains applicable to this certificate and approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate.</td>
<td>None</td>
<td>17/08/2010</td>
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<td>STC No.</td>
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<td>Rev.</td>
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<td>10031486</td>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA STC SA01248SE</td>
<td>PA-31T7</td>
<td>EASA Validation of FAA STC SA01248SE - Installation of two Hartzell HC-44N-3N/D8950SB propellers</td>
<td>None</td>
<td>FAA Supplement AMFS-159SE, Rev. 1: Instructions for Continued Airworthiness Doc., ICA-150-5c, Rev. Air later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CFT (or subsequent revisions after decision)</td>
<td>17/02/2010</td>
<td>Active</td>
<td>10031486</td>
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<tr>
<td>10031482</td>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA STC SA00835SE</td>
<td>PA-31T (CHEYENNE/ CHEYENNE II) PA-31T2 (CHEYENNE III) PA-31T31 (CHEY. V CHEYENNE 6A)</td>
<td>EASA Validation of FAA STC SA00835SE - Installation of two Pratt &amp; Whitney PT6A-135 or PT6A-135A engines and two Hartzell HC-44N-3N/D8950SB propellers</td>
<td>None</td>
<td>FAA Supplement AMFS-157-135A, Rev. 4: Ranger V-1, 135-135A, Revision F, dated 05.08.2010- Continued Airworthiness Doc. ICA-150-5c, Rev. Air later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CFT (or subsequent revisions after decision)</td>
<td>17/02/2010</td>
<td>Active</td>
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<td>10031510</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC SA04-113</td>
<td>BD-700-1A10</td>
<td>Installation; Complete Custom Aircraft Interior - ASN 9349</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ap...</td>
<td>Transport Canada SoC reference C-10-003 dated 21 July 2010-TCCA STC Sa04-113 Issue 1 dated 28 October 2004-Aero Consulting Services Configuration Definition List DB6400 Revision F dated 09 July 2010-or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CFT</td>
<td>20/08/2010</td>
<td>Active</td>
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<tr>
<td>10031514</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH05-4</td>
<td>AS350 B, B1, B2, B3, BA, D</td>
<td>Installation of an Auxiliary Side Locker Fuel Tank.</td>
<td>None</td>
<td>- Master Document List MDM-0350-794, revision D, dated 09.08.2010- Flight Manual Supplement FMS-0350-794, revision C, dated 09.08.2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CFT (or subsequent revisions after decision)</td>
<td>20/08/2010</td>
<td>Active</td>
<td>10031514</td>
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<tr>
<td>10031517</td>
<td>DOA AEGLOBAL LIMITED</td>
<td>FAA TCDS no. J2591</td>
<td>Introduction of Dressen G1A and G4B GatesRev. 1: Transfer from AES Ltd. to new STC holder AES Global Ltd. FAA TCDS No. J2591 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate...</td>
<td>None</td>
<td>None</td>
<td>None</td>
<td>27/04/2021</td>
<td>Active</td>
<td>10031517</td>
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List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
<th>Erickson Aviation Inc. (FAR 23)</th>
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<tr>
<td>Description</td>
<td>Erickson Aviation, Inc.</td>
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<td>Associated Technical Documentation</td>
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<td>Limitations/Conditions</td>
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<td>Status</td>
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<td>System Record</td>
<td>Erickson Aviation, Inc.</td>
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Note: The table provides a list of supplemental type certificates issued by EASA, including the STC holder, certification basis, associated technical documentation, limitations/conditions, date of issue, status, and system record for each entry. The table entries are structured in a tabular format with columns for each of these categories.

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An agency of the European Union
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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
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<th>Models</th>
<th>Description</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<tr>
<td>10031549</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STD SA01065SE</td>
<td>2. FA150, FA155, F152, FA152 2. F150C, F160C, F165C, 2. F160C, F165C, F165J 1. A150E, A150M, 152, A152 1. 150F, 150L, 150M, 150H, 150U 1. 150F, 150L, 150M, 150U</td>
<td>Installation of vortex generators on the wings and vertical tail surfaces on Cessna 151/152 (see applicable Model list)</td>
<td>CS-23.C</td>
<td>Installation of vortex generators on the wings and vertical tail surfaces in accordance with Drawing Package MA201, Revision IR, dated 19th January 2002, and Installation Manual MA082. Revision A, dated 17th April 2002 or later approved revision.</td>
<td>Approval of this change in type design applies to the aircraft listed as MAEL SA01065SE only. This approval should not be extended to other aircraft on which the previous approved modification is incorporated unless it is determined that the relationship between the changes and any other previously approved modification is maintained.</td>
<td>25/09/2010</td>
<td>Active</td>
<td>10031549</td>
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<tr>
<td>10031570</td>
<td>32B SUPPORT SERVICES GmbH</td>
<td>REV. 2</td>
<td>DORNIER 382-100-300</td>
<td>Installation of Emergency Lighting Power Supply System.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this STC:</td>
<td></td>
<td>None</td>
<td>27/09/2018</td>
<td>Active</td>
<td>10031570</td>
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<tr>
<td>10031606</td>
<td>BEECHCRAFT CORPORATION</td>
<td>REV. 2</td>
<td>FAA STC ST109225C-D</td>
<td>MODEL 4000</td>
<td>Installation of Cabin Interior Furnishings and Systems.</td>
<td>The Certification Basis for the original product remains applicable to this STC. The requirements for environmental protection and the associated increases in noise and emissions of the original product are unchanged and remain applicable to this approval.</td>
<td>Only configuration 3, 6, 8 and 9 installations are EASA approved. Airworthiness Limitations inspections are contained in Chapter 4 of the Completion Manual.</td>
<td>16/10/2012</td>
<td>Active</td>
<td>10031606</td>
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<tr>
<td>10031609</td>
<td>TASS Inc.</td>
<td>FAA STC ST00583SE-D</td>
<td>B757-200</td>
<td>Installation of Passenger Accommodations - STC ST00583SE-D</td>
<td>The Certification Basis for the original product remains applicable to this STC. The requirements for environmental protection and the associated increases in noise and emissions of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>None</td>
<td>30/08/2010</td>
<td>Active</td>
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<td>10031612</td>
<td>AERO TECHNICS DESIGN LIMITED</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B737-500</td>
<td>Cabin Seating Recconfiguration from 120Y to 162C/2D layout (A4C)Rev. 1</td>
<td>The Certification Basis for the original product remains applicable to this STC. The requirements for environmental protection and the associated increases in noise and emissions of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously/installed modifications and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/08/2014</td>
<td>Active</td>
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<td>EASA STC no.</td>
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<td>10031614</td>
<td>AERO TWIN, Inc.</td>
<td>FAA STD- SAA4625NM</td>
<td>CESSNA 208, 208B</td>
<td>Fabrication and Installation of Aero TWIN, Inc EBB-200 extended baggagekit in accordance to FAA STC SAA4625NM</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement EBB-200- AFMS, Rev. B dated October 14, 2004</td>
<td>Instructions for Continued Airworthiness EBB-200-DA, last revision dated October 14, 2004 Master Drawing List MOL EBB-200, last revision dated March 6, 2005. All revisions of the above listed documents approved by EASA inc.</td>
<td>None</td>
<td>31/08/2010</td>
<td>Active</td>
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<td>10031622</td>
<td>EMBRAER S.A.</td>
<td>ANAC N *2009S05-07</td>
<td>EMB-500</td>
<td>Installation of a Rockwell Collins Automatic Direction Finder ADF-462/Model</td>
<td>Installation in accordance with Master Document List No TR0083, Rev. J dated 14 May 2009 or later approved revisions. Operation in accordance with Embracer EASA Airplane Flight Manual Supplement (AFMS) for Rockwell Collins ADF-462 ADF system. Rev IP, dated 19 June 2009 or later approved revisions.</td>
<td>- Rotorcraft Flight Manual Supplement No. 17, dated 24th August 2010, other revisions of the above listed documents approved by EASA Design Report DRI 1451222-001 Issue A dated 4th March 2010</td>
<td>1. Required Markings and Placards - In close proximity to the SA-ADF breaker in breaker panel placard &quot;ADF-2&quot;, Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previous modifications will introduce no...</td>
<td>01/09/2009</td>
<td>Active</td>
<td>10031622</td>
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<td>10031633</td>
<td>EUROCOPTER UK Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>MBB-BK 117C2</td>
<td>Installation of a Rockwell Collins Automatic Direction Finder Skyknight System Installation of a Rockwell Collins Automatic Direction Finder Skyknight System with internal cabin control and external loudspeaker array, installed on aircraft RHS and configured with low skid landing gear</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Rotorcraft Flight Manual</td>
<td>- Rotorcraft Flight Manual Supplement No. 17, dated 24th August 2010, other revisions of the above listed documents approved by EASA Design Report DRI 1451222-001 Issue A dated 4th March 2010</td>
<td>none</td>
<td>01/09/2010</td>
<td>Active</td>
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<td>10031637</td>
<td>1263343 ALBERTA INC DBA ENERJET</td>
<td>TCCA STC C- LSA10-077D ISSUE 1</td>
<td>B737-700</td>
<td>Installation of a Rockwell Collins Automatic Direction Finder Skyknight System Installation of a Rockwell Collins Automatic Direction Finder Skyknight System with internal cabin control and external loudspeaker array, installed on aircraft RHS and configured with low skid landing gear</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification Data Summary R1004821 issue A dated April 2010</td>
<td>Modification Review Continued Airworthiness &amp; Safety Improvement R1006846 issue A dated June 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA Design Report DRI 1451222-001 Issue A dated 4th March 2010</td>
<td>Limited to B737-700 aircraft S/N 33920; Maintenance of the Bucher Closet is to be performed by Boeing Aircraft Maintenance/Manual D452/101-CQT(&lt;&gt;) and Bucher Component Maintenance Manual 25-30-65</td>
<td>01/09/2010</td>
<td>Active</td>
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<td>10031638</td>
<td>1263343 ALBERTA INC DBA ENERJET</td>
<td>TCCA STC C- LSA10-311D ISSUE 1</td>
<td>B737-700</td>
<td>Installation of a Rockwell Collins Automatic Direction Finder Skyknight System Installation of a Rockwell Collins Automatic Direction Finder Skyknight System with internal cabin control and external loudspeaker array, installed on aircraft RHS and configured with low skid landing gear</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification Data Summary R0810701 issue C dated April 2010</td>
<td>Modification Review Continued Airworthiness Safety Improvement R1006846 issue A dated June 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA Design Report DRI 1451222-001 Issue A dated 4th March 2010</td>
<td>Limited to B737-700 aircraft S/N 33920; Maintenance of the Recaro seats are to be performed per Recaro Component Maintenance Manual 25-22-11, 25-25-81 and 25-24-79</td>
<td>01/09/2010</td>
<td>Active</td>
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<tr>
<td>Alberta Inc DBA ENERJET</td>
<td>Artex C400-2 ELT Installation</td>
<td>Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificated approval. CR1-12.01 Issue 1: Special Condition - EUROCAE ED Decision 92: The requirements for environmental protection and the associated certificated noise and/or emissions of the original product are unchanged and remain applicable to this certificate/app.</td>
<td>U01/00001/000/000/00/00/00</td>
<td>Limited to B737-700 aircraft S/N 33920 and 34480</td>
<td>02/06/2010</td>
<td>Active</td>
<td>1031863</td>
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<td>TCA STC C-LSA10-251D ISSUE</td>
<td>TIA Model 1545 oven Installation in forward LH monument.</td>
<td>Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificated approval. CR1-12.01 Issue 1: Special Condition - EUROCAE ED Decision 92: The requirements for environmental protection and the associated certificated noise and/or emissions of the original product are unchanged and remain applicable to this certificate/app.</td>
<td>U01/00001/000/000/00/00/00</td>
<td>Limited to B737-700 aircraft S/N 33920 and 34480</td>
<td>01/06/2010</td>
<td>Active</td>
<td>1031864</td>
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<tr>
<td>CAA UK RESPONSIBILITY</td>
<td>Introduction of Swift Broadband</td>
<td>Certification Basis for the original product remains applicable to this certificated approval. CR1-12.01 Issue 1: Special Condition - EUROCAE ED Decision 92: The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/app.</td>
<td>U01/00001/000/000/00/00/00</td>
<td>STC Twenty One Ltd Flight Manual Supplement S21:MS-5060 rev A and later revisions of the above listed documents approved by EASA</td>
<td>02/06/2010</td>
<td>Active</td>
<td>1031862</td>
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<tr>
<td>Bombardier Inc.</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9364</td>
<td>Certification Basis for the original product remains applicable to this certificated approval. CR1-12.01 Issue 1: Special Condition - EUROCAE ED Decision 92: The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/app.</td>
<td>U01/00001/000/000/00/00/00</td>
<td>Transport Canada SoC reference C-10-0310 dated 30 August 2010: TCCA STC No. SA04-113 Issue 1 dated 28 October 2004: Aero Consulting Services Configuration Definition List D685400 Revisions PQ dated 15 August 2010: Instructions for Continued Airworthiness GC92490000-P0CA or later revisions of the above listed documents approved by EASA</td>
<td>02/09/2010</td>
<td>Active</td>
<td>1031863</td>
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<td>Chippewa Aerospace, Inc. Rev. 1</td>
<td>Garmen GTX 330</td>
<td>Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificated approval. CR1-12.01 Issue 1: Special Condition - EUROCAE ED Decision 92: The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/app.</td>
<td>U01/00001/000/000/00/00/00</td>
<td>None</td>
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<td>Lufthansa Technik AG</td>
<td>Trial installation of 2 rows of the BE Aerospace V-seats model BE 15.0PN series 1010/121 as replacement for 3 rows of Recaro EC seat model 341100 Section B of the Lufthansa B747-400 reg DB-ABVW in accordance with HT document STD-000-10/16M Issue 01</td>
<td>Certification Basis for the original product remains applicable to this certificated approval. CR1-12.01 Issue 1: Special Condition - EUROCAE ED Decision 92: The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/app.</td>
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<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD</td>
<td>TCCA SToC-SA10-18</td>
<td>BD-700-1A10</td>
<td>I-CAIR System installation with JetAir Bio-Protection (BPS)/System</td>
<td>EASA 94-A.009</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certifications noise and/or emissions levels from the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC No. SA10-18 Issue 1 dated 09 February 2010. Transport Canada SToC ref. 5010-G-10-0427 dated 31 May 2010. Innotech Aviation Drawing List MDL-BD700-151 Revision NC dated 15 January 2010. Innotech Aviation Airplane Flight Manual Supplement FAMS-010 Revision NC approved 8 February 2010.</td>
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<td>GAMA ENGINEERING LTD</td>
<td>CAA UK RESPONSIBILITY</td>
<td>PIPER PA-34 ALL SERIES</td>
<td>Installation of Garmin G500/G600 PFD/ND</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List MDL1101-01 Issue 1 Change Statement CS1101-01 Issue fInstructions for Continued Airworthiness ICA1101-01 Issue 1</td>
<td>This change is limited to the configuration defined in Change Statement CS1101-01 Issue 1 or later approved revision.</td>
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<td>10031738</td>
<td>AEROSPACE DESIGN AND CERTIFICATION LTD d.T/A AERODAC</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B300</td>
<td>Review of modifications installed prior to Import</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>1) Systems installation and testing in accordance with aeroDAC Form JN222 ADDC (1) Iss 1 dated 28/03/10 Issue 2 Operating Instructions defined in the following Airplane Flight Manual Supplements. a Supplemented Addenda No 1 ADDC (1) Iss 1 dated 31/08/10. Supplement No 002: aeroDAC Form JN222</td>
<td>Aircraft Serial Number FL-327</td>
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<td>10031744</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 1</td>
<td>MYSTERE FALCON 20 B550 Citation II</td>
<td>Enhanced Mode “S” compliance upgrade applicable to Dassault Aviation Mystere Falcon 20 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certifications noise and/or emissions levels from the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Uni Air Enterprise Master Data List MDL1101-10 Revision IR dated 26th August 2010. Uni Air Enterprise Compliance Design Document DEC-011-10 Revision IR dated 23rd August 2010. Uni Air Enterprise Airplane Flight Manual (AFM). Supplement UEAFAIMS-10-002 Revision IR dated 23rd August 2010.</td>
<td>Prior to installation of this modification it must be determined that the international relationship between this modification and any previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10031745</td>
<td>(EASA.A.S.03681)</td>
<td>SAAB ELBE FLUGZEUGWERK E GmbH</td>
<td>A310-324/-338/-324/-325 A300B4-605R/-622R A300B4-601/-603/-620/-622</td>
<td>Counter Stowage installation This STC consists in modifying the Counter Area of aircraft converted by Airbus Pax to Freighter STC's by installing a Counter Stowage</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. CS25 Amendment 4Complementary Condition CC13-1: Crash design conditions/Special Condition CRI S2 Issue 2: Installation of Counter Area AIR 2...</td>
<td>Engineering Order, ref. EFW-E0-0020-001, Issue 8 and later revisions of the above listed documents approved by EASA</td>
<td>The current STC needs specific aircraft configuration as applicable. To install this STC on A310 aircraft models, it must be determined that the international relationship between this modification and any previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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**List of EASA Supplemental Type Certificates**

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<th>Limitations/Conditions</th>
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<td>10031757</td>
<td>BUCHER LEICHTBAU AG (EASA.AS.03984)</td>
<td>REV. 21</td>
<td>SEE AIRCRAFT ALLOCATION TABLE E043-02-12</td>
<td>Installation of B787 (787-8 &amp; 787-9)</td>
<td>The STC has been revised to Installation of Honeywell EGPWS Mark VIII on the following aircraft:</td>
<td>-Flight Manual Supplements - AFMS UAE No.47, Edition 2010.</td>
<td>-Par User Manual 25-67-47 (E017-05-09). This STC constitutes an approval for installation only for aircraft serial numbers as stated in the Aircraft Allocation Table E043-02-12. For additional aircraft serial numbers, the requirements for full compliance have to be demonstrated through a design appraisal.</td>
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<td>10031769</td>
<td>GAMA ENGINEERING LTD.</td>
<td>REV. 3</td>
<td>CAA UK RESPONSIBILITY</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Honeywell EGPWS Mark VIII on the following aircraft:</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Certification Proposal Document CPD-CRS-01, Rev. 3. Certification Plan, ref. doc. No. TN 1988-005, Rev. C - Declaration of Design and Performance - DDP, ref. doc. No. DDP-1888-012, Rev. C - MDL - MDL-CRS-01, Issue 1: ICA: refer to the associated ICA in MDL reference MDL-CRS-01, Issue 1 or later revisions.</td>
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<td>10031773</td>
<td>UNI AIR ENTREPRISE SASUNI AIR ENTREPRISE (EASA.AS.00307)</td>
<td>REV. 1</td>
<td>CESSNA 550 (CITATION X)</td>
<td>Installation of Honeywell EGPWS Mark VIII on the following aircraft:</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EASA TCDS IM A.207 The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List MDL-004-02, Revision 6 from 08 September 2015. Amdt 1 to 2 from 24 May 2004 for S/N 550-0609. AFMS UAE No.15, Edition 2, Revision 2 from 08 September 2015. Amdt 2 to 2 from 24 May 2004 for S/N 550-0916. AFMS UAE No.47, Edition 1, Revision 1 from 14 September 2...</td>
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<td>10031790</td>
<td>TGAS - TOTAL QUALITY AND AIRCRAFTSOLUTIONS ONL, LDA</td>
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<td>A310-304</td>
<td>Cabin Modification to Full Economy Cabin : Reconfiguration from a 151/154 seats configuration</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Design Dossier&lt;&gt; ref. CG-001-015, dated 07 June 2010. Change/Repair Approval Sheet, ref. CG-001-10-MA5 Issue 3, dated 07 June 2010. Service Bulletin, ref. CG-001-10-SRB Issue 0, dated 07 June 2010.</td>
<td>Limited to MSN 573</td>
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<td>INNOTEC AVIATION DIVISION OF IMP GROUP LTD</td>
<td>10031791</td>
<td>FAA STC</td>
<td>REV. 2</td>
<td>TCCA STC No C-LSA10-022/D</td>
<td>B/D-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>EASA TCDS No. IM.A.0007 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EASA STC No. 10031791 TCCA STC No C-LSA10-022/D</td>
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<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>10031793</td>
<td>FAA STC</td>
<td>REV. 1</td>
<td>EC135 T2+, EC135 T3</td>
<td>EC135 T3, EC135 T2, EC135 F1, EC135 P2, EC135 P2*</td>
<td>Cable Cutters Kit Installation of Apical Industries, Inc. Cable Cutters Kit welds to fabric on the inside of the aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC SR20350LA REV. 1: Amendment of the STC in order to add the applicability to EASA STC is based.</td>
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<td>AMSAFE, Inc.</td>
<td>10031808</td>
<td>FAA STC</td>
<td>REV. 2</td>
<td>ST01512LA</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of AmSafe Seatbelt Airbag System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements-Special Condition (s) CRI D-01 “HIC Compliance Inflatable Restraint System” and CRI F-116 “Non-rechargeable Lithium Battery Installations.” The requirements for environmental protection and any emissions level of the product are unchanged and remain applicable to this certificate/approval.</td>
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<td>IXE AEROSPACE, Inc. - FSI</td>
<td>767-300</td>
<td>Installation of an Interior Reconfiguration in A330-300 Series aircraft</td>
<td>In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>- FAA STC 767-300/BSE dated 24th December 2008 - Master Drawing List ref. doc. No. ABF827-D01, Revision P, dated 7th July 2010 - Airplane Flight Manual Supplement, ref. doc. No. TP16-827AFM, Revision1, dated 6th January 2010 - Supplemental Instruction for Continued Airworthiness, ref. doc. No.ABF827-D.0...</td>
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<td>FIELD AVIATION COMPANY Inc.</td>
<td>AS 332 L, AS 332 L1</td>
<td>Installation of 4 Place Auxiliary Seat</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Tank Installation 300151-3-6346 revision D dated 14 October 1987 - Drawing List E300151 revision G dated 9 September 1987 - Master Drawing - FACI-6346-02 amendment 1 dated 6 November 1987 - Maintenance Manual Supplement...</td>
<td>None</td>
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<td>DREF STIFTUNG LUFTRETTUNGE EMERGENZUGZE AG</td>
<td>MB8-BK17B-2, MB8-BK117 C-1</td>
<td>Installation of second directional gyro system installation of a second directional gyro system for IFR operation based on DRE Major Change Project 240 and defined in MD-117-34-20-00001.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Flight Manual Supplement 15-DRF-05, Rev. 1 or later revisions of the above listed documents approved by EASA - Installation Manual IM-117-34-20-00001, Rev. 0b - Instruction for Continued Airworthiness CA-117-34-20-00001, Rev. A - Master Document List MD-117-34-20-00001, Rev. A or later revisions of the above.</td>
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<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>767-300</td>
<td>Cabin reconfiguration - Air Seychelles 767-300 aircraft reconfiguration by replacing the existing J-class seat with an electrically powered seat. There are adaptations to the existing PSU. Y-class cabins to suit new layout Rev. 4 - Transfer of STC from SR Technics Switzerland to groWNEngineering Par...</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate.</td>
<td>- Change Classification and Certification Sheet doc. Ref. C767-25-1010Rev 03 dated 18 November 2010 - Master Data List doc. Ref. C767-25-1010-MLX Rev 03 dated 18 November 2010 - Instructions for Continued Airworthiness doc. Ref. C767-25-1010-CA-A01Rev 01 dated 08 Nov 2010 or later revisions of the above.</td>
<td>None</td>
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<td>10031836</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>A330-300</td>
<td>Update of a Panasonic System e2 In-Flight Passenger Entertainment System</td>
<td>In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>- FAA STC ST02755NY dated 9th January 2010 - Master Data List ref. doc. No. 0514-0050-2200, Revision B, dated 30th December 2009 - Electrical Wiring Interconnection System (EWIS) Assessment, ref. doc. No. 6654-0706-2200, Revision - dated 1st July 2010 or later revisions of the above listed documents...</td>
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<td>The Certification Basis for the original product and the associated certified noise level...</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9379</td>
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</table>

List of EASA Supplemental Type Certificates

easa.europa.eu

Airbus Corporate Jet Centre Modification CJ0540 Issue 4 - VIP Interior /QATARI Revision due to update of AFM-S and MPD-S

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements as listed in the General ORI A-2301-007 "AGCJ Project Certification Basis". The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification.

This STC is limited to A320-232 MSN 4170 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Installation of Northrop Grumman Dual Installation, Complete Custom Aircraft

The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the installation of the associated certified noise level.

None

This STC is limited to A320-232 MSN 4170 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Installation of an International Aircraft Systems LITEF LCR-100 AHRS/FAASTC ST02025LA-D (issued March 10, 2010)

The Certification Basis for the original product and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification.

Previous or concurrent requirement for this installation is approved installation of the LCR-93 AHRS in accordance with thestalj EDR 4487.

Installation in accordance with Scandinavian Avionics Design document FM-S968325 STC (STC Document List) Operation in accordance with Scandinavian Avionics Design document nF968325-AM (Aircraft Flight Manual Supplement) Instructions for Continued Airworthiness in accordance with ScandinavianAvionics...

Installation, Complete Custom Aircraft Interior - ASN 9379

EASA TCDS No IMA-0097 The Certification Basis for the original product remains applicable to the installation of the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification.

Restricted to ASN 9379

* An agency of the European Union
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<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
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<td>10031901</td>
<td>WORLDWIDE AIRCRAFT SERVICES Inc.</td>
<td>FAA STC SA48725W</td>
<td>SA226-T, SA226-AT</td>
<td>Installation of 2 Garrett AiResearch engines Model TPE 331-10UA-501G or TPE 331-10UA-511G</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. CAR 3</td>
<td>STC Drawing list. AF-294, Rev. E, dated September 1, 2010. Installation instructions: D-10273, Rev. C, dated September 3, 2010. Installation location instruction: (1 place, 3 place divan). AF-497, AF-498, AF-499, etc.</td>
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<td>10031903</td>
<td>AVIATION SERVICE A.S.</td>
<td>407</td>
<td>FAA STC SA48725W</td>
<td>Installation of alternate static port for VFR night operation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FM 30 30/01 001 revision 0 dated 24 August 2010 - Flight Manual Supplement - alternate static port</td>
<td>None</td>
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<td>10031917</td>
<td>AVIATION FABRICATORS, Inc.</td>
<td>FAA STC SA2818SW</td>
<td>441</td>
<td>Installation of 1, 2 or 3 place side seating divans in accordance with FAA STC SA2818SW</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>STC Drawing list: AF-294, Rev. E, dated September 1, 2010. Installation instructions: D-10273, Rev. C, dated September 1, 2010. Installation location instruction: (1 place, 3 place divan). AF-497, AF-498, AF-499, etc.</td>
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<td>10031933</td>
<td>ASI INNOVATION</td>
<td>T206H</td>
<td>ASI-fam-0009-01</td>
<td>Installation of FLIR MX10 mechanical provision on T206Haircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Limitation to Cessna T206H serial number 445 and 446</td>
<td>23/09/2010</td>
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<td>MECAER AVIATION GROUP S.p.A.</td>
<td>AW139 / A8139</td>
<td>IFEEL Entertainment System Installation PIN 6AB1AD016-001</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- 6AB1RAD-176 Rev. C. IFEEL System Description- 6AB1RM-2-30A Rev. A</td>
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<td>10031942</td>
<td>AIR METHODS CORPORATION</td>
<td>FAA STC SR00702DE-D</td>
<td>BELL 429</td>
<td>Installation of Supplemental Strobe Lights</td>
<td>The Certification Basis for the original product remains applicable to the already approved. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval</td>
<td>EASA STC 063/2010 dated 11 March 2010</td>
<td>None</td>
<td>23/02/2010</td>
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<td>ACS-NAI Ltd.</td>
<td>TCCA STC #C-LSA10-034/D 1SS-1</td>
<td>BD-700-1A10</td>
<td>Installation of Two 16G Side Facing Divans</td>
<td>EASA TCDS 06016260C The Certification Basis for the original product remains applicable to this installation of the Goodrich De-Icer boots. The requirements for environmental protection and the associated certified noise levels remain unchanged and remain applicable to this certificate/approval</td>
<td>Transport Canada CAA ref. C-10-0708 dated 22 September 2010</td>
<td>Restricted to ABN 93/47</td>
<td>23/02/2010</td>
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<td>10031946</td>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA STC SA10816SC</td>
<td>425</td>
<td>Installation of Ultra Electronics engine controller modification in accordance with FAA STC SA10816SC</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this installation of the Goodrich De-Icer boots. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval</td>
<td>Installation Drawing No. 290725-50, Revision IR, dated July/4, 2007</td>
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<td>GOODRICH CORPORATIONS ENSORS AND INTEGRATED SYSTEMS</td>
<td>TCCA STC SA10-3</td>
<td>DHC-8-401, DHC-8-402</td>
<td>Installation of Goodrich De-Icer boots on the Aircraft Wings, Horizontal Stabilizer and Vertical Stabilizer/FAA STC ST02284NY, revised September 29, 2010 (identical with TCJA STC SA10-3, issue 2)</td>
<td>JAR/CS-27, 23.627, 23.771, 23.1183...</td>
<td>Installation of the Goodrich De-Icer boots must be in accordance with TCJA approved “Goodrich Corporation Installation and Maintenance Instructions, Pneumatic Ducts for DHC-8 Q 400” Report ERR-2010-014 Revision B, dated June 03, 2010. Maintenance of the Goodrich De-Icer boots must be in accordance with the above listed documents approved by EASA.</td>
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<td>BOND HELICOPTERS EUROPE Ltd.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>AS 332 L2</td>
<td>Honeywell TCAS II Installation in accordance with BOND/MOD/019Rev.1 new Build Standard BD/18-100-06 included Rev. 2, installation of TCAS II Equipment with logic version 7.1</td>
<td>The Certification Basis for the original product as amended by all subsequent additional or alternative airworthiness requirements - Special Condition: CNR F-01 “Protection from the effects of HIRF”, dated 21 September 2010. The following paragraph(s) apply at a later amendment: CS 29 Amendment 2, dated 17 N...</td>
<td>For TCAS II with logic version 7.1, Flight Manual Supplement FM/019-001, 022 “Honeywell TCAS II installation” Issue 1 For TCAS II with logic version 7.1, Flight Manual Supplement FM/019-017 “Honeywell TCAS II installation with ACAS II level” Issue 7.1, Flight Manual Supplement FM/019 of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>Activation of Honeywell Runway Awareness &amp; Advisory System (RAAS)</td>
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<td>ARMSTRONG AEROSPACE Inc.</td>
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<td>Customized Aircraft Completion - ASN 9365</td>
<td>EASA M.A.2009</td>
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<td>Installation of an Alt Lavatory Compartiment.</td>
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<td>FAA STC ST02376CH</td>
<td>ATR 72-201, ATR 72-202, ATR 72-212</td>
<td>Revision of EASA STC 10032016: Removal of the G1, G2 &amp; G5 galley systems established on initial issue of EASA STC 10032016 date of issue 01/10/2010 granted to MacCarthy/Interiors Ltd Revision: Transfer of Certificate to Magnetic MRO AS</td>
<td>Installation of an O2 Rack</td>
<td>MSN4114 Interior Outfitting: The change consists in a complete cabin interior outfitting with a 28passenger VIP interior including two medical stretcher units. Major Change to EASA STC 10032043 Certification for commercial operation</td>
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<td>MSN4114 Interior Outfitting: The change consists in a complete cabin interior outfitting with a 28passenger VIP interior including two medical stretcher units. Major Change to EASA STC 10032043 Certification for commercial operation</td>
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**List of EASA Supplemental Type Certificates**

**Certification Basis**

The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval:

- Revision B of FAA STC ST0376CH (as amended by FAA on 24th June 2010) regarding installation of a Rockwell Collins HF-9000 HF Radio System in accordance with Armstrong Aerospace Master Data List, Doc. Number DC104-4033-00 Revision E, dated 27th May 2010, or later approved revisions.

**Limitations/Conditions**

- Approval limited to MSN4114 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

**Date of Issue**

- 30/06/2010

**System Record**

- 10032001

**Status**

- Active
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<td>LUFTHANSA TECHNIK AG</td>
<td>10032061</td>
<td>REV. 1</td>
<td>STDD34-10/01M</td>
<td>MD-11</td>
<td>Activation of ADS-B Out capability via MD-11</td>
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<td>ALTITUDE AEROSPACE INTERIORS Ltd.</td>
<td>10032064</td>
<td>REV. 2</td>
<td>SDC-AFMS-0703</td>
<td>747-400</td>
<td>Installation of Honeywell / Racial MS6000 SatCom System per Altitude Aerospace Interiors Master Data List MDL 22001052 Rev. B Rev. 2 to include aircraft serial number</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List MDL 22001149, Altitude Assurance Report SR 21001439, AICA reference: TR 35001213 Rev B</td>
<td>Prior to installation of this modification it must be determined whether the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>10/07/2015</td>
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<td>10032056</td>
<td>GFD GmbH</td>
<td>10032066</td>
<td>REV. 1</td>
<td>LSA10-43</td>
<td>35A, 36A 35A / 36A</td>
<td>COMJAM System Installation Rev. 1: Transfer to GFD GmbH</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval Refer to Certification Program 856-049, rev. IR: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List 856-049, Rev. A REV.</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>10032067</td>
<td>REV. 1</td>
<td>LSA10-1A11</td>
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<td>EASA IM.A.023: The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>EASA Master Document List No. 078R-MDL Revision N/C dated 04 October 2010 RUAG AFMS No. 078R-AFMS Revision N/C dated 17 August 2010 RUAG SuC ref. 079R-CH Revision N/C dated 30 September 2010 or later revisions of the above listed documents approved by EASA</td>
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<td>EASA IM.A.023: The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>EASA Master Document List No. 078S-MDL Revision N/C dated 19 September 2010 RUAG AFMS No. 078S-AFMS Revision N/C dated 01 October 2010 RUAG SuC ref. 078S-CH Revision N/C dated 04 October 2010 or later revisions of the above listed documents approved by EASA</td>
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<td>DECA AVIATION ENGINEERING Ltd/SHA PAL AEROSPACE</td>
<td>10032069</td>
<td>REV. 1</td>
<td>LSA10-43</td>
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<td>Radar Antenna Installation - Structural Provisions a) TCCA STC O-LSA10-43 b) DHC-6-300</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
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The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval. | none                        | none            | 06/10/2010 | Active     | 10032064 |
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<td>BD-700-1A10</td>
<td>Sound Reduction Kit</td>
<td>EASA MAA.029. The Certification Basis for the original product remains applicable to this certificate/approval The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval</td>
<td>FAA SOC dated 10 September 2010, FAA STC No. ST02675AT, dated 02 April 2004, AIMS International Corp. Engineering Top Level Training SG02521522 Rev IR dated 15 March 2003 and installation instruction document No. AIMS-INST-1 Rev IR dated 16 April 2003 or later approved revisions. AIMS International Corp.</td>
<td>None</td>
<td>06/10/2010</td>
<td>Active</td>
<td>1022509</td>
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<tr>
<td>ALTITUDE AEROSPACE INTERIORS Ltd.</td>
<td>B747-400</td>
<td></td>
<td></td>
<td></td>
<td>Refurbishment of the B747-400 Upper and Main deck interior installation and seating configuration in accordance with Altitude Aerospace Interiors Master Data List MDL 22001038 Rev. A or later EASA approved/revision</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval. CRI D-01 Width of Aisle front seats Zone A-CRI D-02 Inflatable Restraint fro Head Injury Protection, CAA-UK Special Condition CAIA-SC-A-03.</td>
<td>Instructions for Continued Airworthiness Altitude report TR35001213 Rev. B shall be incorporated in the airplane maintenance documentation later revisions of the above listed documents approved by EASA</td>
<td>This STC is applicable to A/C Serial Numbers 24855, 24996, 24997, 25665, 26910, 27692 and 29375. Several parts installed with this STC deviate from Part 21A.604 regarding EPA marking. The responsible NAA grant a 14.4 exemptions to accept the aircraft on its register. As this will be a permanent exemption...</td>
<td>06/10/2010</td>
<td>Active</td>
<td>1022509</td>
</tr>
<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST04204AT-D</td>
<td>REV. 1</td>
<td></td>
<td>GV-SP</td>
<td>Installation of the Aircraft Cabin Compartment for Gulfstream GV-SP Serial Number 5284.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. JAR 25 Amd 1STThe requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List No. GCS20087102 Rev F dated September 27, 2010. Operation in accordance with Gulfstream Aerospace Corporation Aircraft Flight Manual Supplement No. GCS20087145 Rev. dated September 27, 2010. Instructions for Continued Airworthiness Altitude report TR35001213 Rev. B shall be incorporated in the airplane maintenance documentation later revisions of the above listed documents approved by EASA.</td>
<td>Aircraft Serial Number 5284 Only.</td>
<td>23/02/2012</td>
<td>Active</td>
<td>1032074</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST03793AT-D</td>
<td>REV. 1</td>
<td></td>
<td>GV-SP</td>
<td>Installation of an Airplane Interior for aircraft Serial Number 5277</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Install in accordance with Master Index List GCS15491527, Revision C, dated August 27, 2010. To be operated in accordance with FAA approved Airplane Flight Manual Supplement, GCS15468M014, Rev. **. dated August 26, 2010 or later revisions of the above listed documents approved by EASA/inaccordance with...</td>
<td>This approval is limited to only the installation made in Gulfstream GV-SP, AC 25.527. The following STC’s are a required part of this STC: ST02698AT-D, ST02967AT-D, ST03533AT-D, ST07297AT-D, ST07271AT-D, and ST03296AT-D. It is the installer's responsibility to ensure these STC’s are EASA approved.</td>
<td>06/10/2010</td>
<td>Active</td>
<td>1032075</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST02675AT</td>
<td>REV. 1</td>
<td></td>
<td>GV-SP</td>
<td>Installation of the 115V AC/60Hz Outlets Multiple STC for GulfstreamGV-SP model aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. JAR 25 Amd 1STThe requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List No. GCS16496000 Rev D, dated May 30, 2008. Operation in accordance with Gulfstream Aerospace Corporation Aircraft Flight Manual Supplement No. GCS14149M000 Rev. dated November 17, 2006 (-401), or GCS14149M002 Rev A, dated May...</td>
<td>none</td>
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<td>EASA STC No.</td>
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<td>FAA STC or STC No.</td>
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<td>Foreign/NAA STC No.</td>
<td>Aircraft Model</td>
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<td>Limitations/Conditions</td>
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<td>10032077</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST03794AT-D</td>
<td>REV. 1</td>
<td></td>
<td>GV-SP</td>
<td>Installation of the Aircraft Cabin Compartment for Gulfstream GV-SP model aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/app/</td>
<td>None</td>
<td>23/02/2012</td>
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<td>10032080</td>
<td>AEROLITE AG</td>
<td>EMS Equipment PIN 139200-501</td>
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<td>Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC52008695 Rev 6, dated September 3, 2010.</td>
<td></td>
<td>None</td>
<td>06/10/2010</td>
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<td>10032086</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST03794AT-D</td>
<td>REV. 1</td>
<td></td>
<td>GV-SP</td>
<td>Installation of the Audio Video Distribution System (AVDS) Multiple STCs for Gulfstream GV-SP model aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/app/</td>
<td>None</td>
<td>23/02/2012</td>
<td>Active</td>
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<td>10032094</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>FAA STC NO. ST00411LA</td>
<td></td>
<td></td>
<td>B747-400</td>
<td>Installation of Honeywell Traffic Alert and Collision Avoidance System (TCASII).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/app/</td>
<td>None</td>
<td>07/10/2010</td>
<td>Active</td>
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<td>10032098</td>
<td>CAA UK RESPONSIBILITY</td>
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<td>BOMBARDIER CL600-2B19</td>
<td>Removal of Divan Seating Restriction</td>
<td>EASA JMA-023</td>
<td>The Certification Basis for the original product remains applicable to this certificate/app/</td>
<td>None</td>
<td>07/10/2010</td>
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<td>10032104</td>
<td>JAMCO AMERICA, INC.</td>
<td>FAA STC ST01094SE</td>
<td></td>
<td></td>
<td>747-400</td>
<td>Installation of an interior reconfiguration of the main endupper deckin 747-400 airplanes in accordance with Jamco America, Inc. Master Drawing List no. IA-072MDL, Revision AD dated 19 September 2008 or later approved revision/Validation of an FAA STC ST01094SE.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/app/</td>
<td>None</td>
<td>07/10/2010</td>
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<tr>
<td>Honeywell International Inc.</td>
<td>MF900 &amp; F900EX</td>
<td>230VAC electrical distribution for galley. An inverter and a GFI are installed to provide 120VAC power on one outlet in the galley.</td>
<td>Certification Basis for the original product remains applicable. All revisions of the above listed documents approved by EASA.</td>
<td>EASA STC no. REV.04-5200/725, Amdt. 10.10.2010.</td>
<td>none</td>
<td>13/10/2010</td>
<td>Active</td>
<td>10032163</td>
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<td>Dassault Falcon Service</td>
<td>PA-32-301TC</td>
<td>Dual/single installation of Garmin GNS430(A)/GNS530(A)W</td>
<td>Certification Basis for the original product remains applicable. All revisions of the above listed documents approved by EASA.</td>
<td>EASA STC no. REV.04-5200/725, Amdt. 10.10.2010.</td>
<td>none</td>
<td>18/10/2010</td>
<td>Active</td>
<td>10032179</td>
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<td>Airbus Transport International SNC</td>
<td>A300F4-60BST</td>
<td>Special Transports / Transports SpéciauxTris STC, ref. ST25MOD020, ed.01 (4)</td>
<td>Certification Basis for the original product remains applicable. All revisions of the above listed documents approved by EASA.</td>
<td>EASA STC no. REV.04-5200/725, Amdt. 10.10.2010.</td>
<td>none</td>
<td>15/10/2010</td>
<td>Active</td>
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<td>Dassault Falcon Service</td>
<td>F900EX</td>
<td>115/230VAC distribution circuit in the cockpit to switch off all power outlets in the cockpit.</td>
<td>Certification Basis for the original product remains applicable. All revisions of the above listed documents approved by EASA.</td>
<td>EASA STC no. REV.04-5200/725, Amdt. 10.10.2010.</td>
<td>none</td>
<td>15/10/2010</td>
<td>Active</td>
<td>10032204</td>
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Notes:
- EASA STC no. REV.04-5200/725, Amdt. 10.10.2010. refers to the certification basis for the original product and all subsequent revisions.
- Certification Basis for the original product remains applicable. All revisions of the above listed documents approved by EASA.
- Modification applicable to F900EX S/N 086 with a pre modification configuration features a 12-passenger and 3-crew accommodation.
<table>
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<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAATC ref.</th>
<th>Model</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tr>
<td>10032214</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>A320-232</td>
<td></td>
<td>Cabin Interior Reconfiguration and In Flight Entertainment (IFE) Modifications on Etihad Airways A320-232 Aircraft</td>
<td>Certification Basis as per EASA TCDS A.064The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CS-25 Appendix H and AMC Appendix H25. If the requirements for environmental protection and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- NAT Certification Summary, ref. doc. N0362-C00, Revision 0, dated 17th October 2010- NAT Master Drawing List, ref. doc. N0362-D10, Revision 0, dated 15th October 2010</td>
<td>Not applicable</td>
<td>18/10/2010</td>
<td>Active</td>
<td>10032214</td>
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<td>10032216</td>
<td>SABENA TECHNICS BOD</td>
<td>F28 MK0100</td>
<td></td>
<td>Modification wireless telephones installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification plan DO-Nr.2009-174-PDE, issue 2, dated 15th September 2010</td>
<td>None</td>
<td>18/10/2010</td>
<td>Active</td>
<td>10032216</td>
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<tr>
<td>10032232</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPO NSIBILITY</td>
<td>CESSNA 206H, T206H</td>
<td>Introduction of Stand-By Battery Power Supply in accordance with CS21.900-970, issue 2, dated 14/10/2010</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement S21.FMS-0070, issue 1</td>
<td>None</td>
<td>19/10/2010</td>
<td>Active</td>
<td>10032232</td>
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<td>10032241</td>
<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>FAA STC SA03860AT</td>
<td></td>
<td>Honeywell KN-62A DME</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Honeywell KN-62A DME Instructions for Continued Airworthiness X21.3453-61-0002AFMS X21.3453-61-1012 Rev IR dated July 06 2010 or later revisions of the above listed documents approved by EASA</td>
<td>none</td>
<td>05/11/2010</td>
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<td>10032246</td>
<td>BEECHCRAFT CORPORATION</td>
<td>FAA STC ST105665C-D</td>
<td>4000</td>
<td>Installation of Flight Camera System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>STC Required Documents List (RDL) No. 240-2002-1, revision 11, October 13, 2009 Complete Manual Chapter 44-20-30-H4000, August 2009 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>19/10/2010</td>
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<td>10032246</td>
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<td>10032248</td>
<td>BOMBARDIER INC.</td>
<td>REV. 1 TCCA STC NO. SA06-60</td>
<td>BD-700-1A11</td>
<td>Modification to incorporate changes to the Installation of Custom Airframe Interiors - ASN 9370. Revision to EASA STC 10032248</td>
<td>EASA TCDS No. 9M.A.009The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-.TCCA STC No. SA06-60 Issue 1 dated 14 June 2005-TCCA SuC ref. C-10-002 dated 1 November 2010 -Aero Consulting Services Ltd. Configuration Definitions List E025000 Revision DY dated 28th October 2010-Bombardier Aerospace Airplanes Flight Manual Supplement G- FMS2504501Revision NIC or later approved rev...</td>
<td>Restricted to ASN 9370</td>
<td>09/11/2010</td>
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<td>Associated Technical Documentation</td>
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<td>1003253</td>
<td>DART AEROSPACE Ltd.</td>
<td>2</td>
<td>TCCA STG SH09-37 ISSUE 1</td>
<td>D412-791 Safety Enhancement Kits</td>
<td>The Certification Basis for the original product remains applicable to the STC approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FMS-D052-791 rev A; FMS-D142-791 rev A; FMS-D122-791 rev A; ICAD-412-791 rev 0; MDL-412-791 rev A; MIL-D-412-791 rev 0.</td>
<td>None</td>
<td>03/07/2024</td>
<td>Active</td>
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<td>1003264</td>
<td>AIRCRAFT TECHNICAL DEVELOPMENT Inc.</td>
<td>1</td>
<td>FAA STC ST02317LA</td>
<td>A330-200 Installation of Cabin Interior Noise Reduction Kit</td>
<td>In accordance with the EASA TCDSThe Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the STC:</td>
<td>FAA STC STG137LA, dated 6th October 2010.</td>
<td>Not applicable</td>
<td>21/10/2010</td>
<td>Active</td>
<td>1003264</td>
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<td>1003268</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>REV. 3</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>EC135 T2+, EC135 P2+</td>
<td>NVIS Certification EC135 P2+ / EC135 T2+ Note 1: Rev 1 of EASA STC No. 1003268 is raised in order to modify the STC as follows: 1. Extend applicability from EC135 T2+ to EC135 P2+ model. 2. Introduce a third Night VI.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- RFMS EC135 Supplement 56, Issue 2 (SINGLE PILOT OPERATIONS TO 5000ABOVE GROUND LEVEL NVIS COCKPIT LIGHTING - NIGHTVISIONGOGGLES) for configurations controlled by ICAO/AS12142 documents. - RFMS AHUK/135P-103 Rev 0 II (EC 135 NVIS OPERATIONS TO GROUND LEVEL (to Ground Level)) for configurations controlled by...</td>
<td>a) Aircraft Build Standard Prior to installation of either AHUK options, the aircraft must have the following AHD factors: at least 2 x NVG Friendly External Lighting Kit - L3312-003-00; Cockpit Layout NVG Standard - L3311-002-211. NVG Friendly Cabin and Cargo CompartmentAHU Option...</td>
<td>29/11/2016</td>
<td>Active</td>
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<td>1003277</td>
<td>FIBERGLAS-TECHNIK RUDOLF LINDEMANN &amp; CO. KG</td>
<td>REV. 3</td>
<td>GROB G 163 &quot;TWIN II&quot;</td>
<td>STC &quot;Reparatur Twin II&quot;</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>STC &quot;Reparatur Twin II&quot; - Flight manual prepared for the installation of either AHUK options.</td>
<td>Even if parts from a &quot;Twin II&quot; without Acro are used, the combined airplanefly will be always a &quot;Twin II&quot; without Acro.</td>
<td>22/10/2010</td>
<td>Active</td>
<td>1003277</td>
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<td>1003278</td>
<td>FLYING COLOURS CORP.</td>
<td>REV. 1</td>
<td>TCCA STC NO. SA01-68</td>
<td>Installation, Equipment Bay Catwalk</td>
<td>EASA M.A 023 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>TCCA SaC ref. C-10-0781 dated 07 October 2010. TCCA STC No. SA10-46 dated 31 August 2010. ACS NAVI Ltd Modification Data Summary GH44000 Revision A. FLYING Colours Corp. Equipment Bay Catwalk Instructions for Continuation/Interimness, GH4401 Revision NG dated...</td>
<td>None</td>
<td>28/10/2010</td>
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<td>FLYING COLOURS CORP.</td>
<td>REV. 1</td>
<td>TCMA STC NO. SA10-67</td>
<td>CL-600-2B19</td>
<td>Installation, Equipment Bay Ladder</td>
<td>EASA MA.023 The Certification Basis for the original product remains applicable to this certificate's approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate.</td>
<td>TCMA STC No. C-10-4770 dated 07 October 2010; TCMA STC No. SA10-67 Issue 1 dated 31 August 2010; ACS-NA4 Ltd. Modification Data Summary G745600 Revision A; Flying Colours Corp. Equipment Bay Ladder Stowage Provisions Instructions for Continued Airworthiness, G745071 Revision NC or later revisions of the ab. Prior embodiment of EASA STC No. 10032278 Installation, Equipment Bay Catwalk is required.</td>
<td>28/10/2010</td>
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<tr>
<td>FLYING COLOURS CORP.</td>
<td>REV. 1</td>
<td>TCMA STC NO. SA10-65</td>
<td>CL-600-2B19</td>
<td>Installation, Equipment Bay Access Door Steps</td>
<td>EASA MA.023 The Certification Basis for the original product remains applicable to this certificate's approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate.</td>
<td>TCMA STC No. C-10-4779 dated 07 October 2010; TCMA STC No. SA10-65 Issue 1 dated 01 September 2010; ACS-NA4 Ltd. Modification Data Summary, G745600 Revision A; Flying Colours Corp. Equipment Bay Access Door Steps Instructions for Continued Airworthiness, G745071 Revision NC or later revisions of the ab.</td>
<td>none</td>
<td>28/10/2010</td>
<td>Active</td>
<td>10032286</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td>SF340A AND 340B</td>
<td>Installation of Collins TCAS-94 system with dual Collins TDR-94D transponders.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate's approval: CS-25, am 40(CR) H-01, Issue 2. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate.</td>
<td>Engineering Order and Certification Program 2055179-EO, Issue 1; List of Certification Documents 2055179DC, Issue 1; List of Installation Documents 2055179STC, Issue 1; AFM-supplement 2055179-AM-A, Issue 2; AFM-supplement 2055179-AM-B, Issue 2 or later revisions of the listed documents approved.</td>
<td>Transponder system NOT approved for ELIs or EHS.</td>
<td>28/10/2010</td>
<td>Active</td>
<td>10032285</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td></td>
<td>FAA STC SA9105W-D</td>
<td>BEECH S35, V35, V35A AND V35B</td>
<td>Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-577 (14 Volt System)</td>
<td>The Certification Basis for the original product remains applicable to this certificate's approval. The certificate's noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate's approval.</td>
<td>FAA-Approved S-TEC Bulletin No. 677, dated 04-Dec-09; Revision 6; later revisions approved by FAA-Approved S-TEC Master Drawing List No. 92756, dated 04-Dec-09; Revision F. or later approved revision FAA-Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N 891151; Revision 2.</td>
<td>This STC is also eligible on Beech models S35, V35, V35A, and V35B when modified by STC SA5223NM (turbocharged engines) or STC SA5223NM (turbocharging system) and any combination of the following STCs: STCSA090742C (vortex generators), STC S4-1629 (tip tanks).</td>
<td>22/10/2010</td>
<td>Active</td>
<td>10032286</td>
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<tr>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td></td>
<td>A330-200/-300</td>
<td>Air Ambulance Equipment &quot;quick conversion&quot; (P/N A330-25-20-10.000 515 - Intensive Care Module P1 A330-25-20-10.300-603)</td>
<td>The Certification Basis for the original product remains applicable to this certificate's approval. The certificate's noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate's approval.</td>
<td>Compliance Record, ref. doc. CR-A330-10.000-515, Revision 0, dated 25th October 2010; Component, Installation and Operation Manual, ref. doc. CMM- A330-ATT73, Rev. 1, dated 06 August 2010; Aircraft Flight Manual Supplement, ref. doc. AMT-54, Rev. 7 or later revisions of the listed documents approved.</td>
<td>This STC does not constitute an approval for installation Compliance with the following requirements will be demonstrated and covered by the DOA responsible for the medical stretcher installation on the aircraft: A330 through a design approval § 25.501(d) §§ 25.785(b), 25.785(c), §§ 25.785(j), 25.785(k), § 25.791(i), 25.80...</td>
<td>25/10/2010</td>
<td>Active</td>
<td>10032292</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>Movie</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10032296</td>
<td>865 AEROSPACE LIMITED</td>
<td>REV. 2</td>
<td>TC1A-2C ref.</td>
<td>LSA09-0210</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>TCCA ST03881AT - LSA09-0210 Issue 3 dated 17 June 2013 ACS-NL Ltd. Modification Data Summary G116000 Revision F.Aero Consulting Services Ltd. Mandatory Maintenance Manual Supplement G116070 Revision 10, Savannah Air Center Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior</td>
<td>Restricted to ASN 8945</td>
<td>16/02/2013</td>
<td>Active</td>
<td>10032296</td>
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<tr>
<td>10032318</td>
<td>365 AEROSPACE DDS LIMITED</td>
<td>REV. 2</td>
<td>DHC-8-401, DHC-8-402</td>
<td>Installation of Ground Recognition LightsRev. 2: Transfer of STC from 365 Aerospace Ltd. to 365 Aerospace DDS Ltd.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 8 for changed areas. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aerospace Design &amp; Engineering Consultants Ltd. Manual Supplement No. ADSM-DHC-8-400-FLYBE, dated 01 August 2011or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effects upon the airworthiness of the product.</td>
<td>05/08/2019</td>
<td>Active</td>
<td>10032318</td>
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<tr>
<td>10032322</td>
<td>SAVANNAH AIR CENTER, L.L.C.</td>
<td>REV. 1</td>
<td>FAA STC ST03881AT</td>
<td>Installation, Complete Custom Interior, S/N 9345 Configuration</td>
<td>EASA TCDS No IM-A-0097: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>FAA SoC ref. ST12210AT-T dated 21 October 2010 FAA STC No. ST03881AT dated 26 October 2010 Savannah Air Center MOD. XRS9345-MDL-S850001. Rev. A dated 13 September 2010.</td>
<td>Restricted to AS 9460 and 1. Refer to the limitations sections of the referenced FCOMS:G116000 Rev. A - FCOMS - MDL-250001; Rev. A dated 21 October 2010.</td>
<td>29/03/2011</td>
<td>Active</td>
<td>10032322</td>
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<tr>
<td>10032327</td>
<td>BOMBARDIER INC.</td>
<td>REV. 1</td>
<td>TC1A STC NO. SAV04-113</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9340</td>
<td>EASA TCDS No IM-A-0097: The Certification Basis for the original product is applicable only to IEEE 802.11(b) WLAN system.</td>
<td>Transport Canada SoC reference C-10-0242 dated 21 October 2010 TC1A STC No. SAV04-113 Issue 1 dated 28 October 2004.</td>
<td>Restricted to AS 9360 and 1. Refer to the limitations sections of the referenced FCOMS:G116000 Rev. A - FCOMS - MDL-250001; Rev. A dated 21 October 2010.</td>
<td>27/10/2010</td>
<td>Active</td>
<td>10032327</td>
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<tr>
<td>10032329</td>
<td>NW5 CERTIFICATION SERVICES Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING B747-400</td>
<td>Installation of SaftGlo Inc. Photoluminescent Floor Proximity Emergency/Escape Path Marking System (FPEEPMS) in accordance with NW5Certification Services Design Report 3046-CMS-01 Rev. 2 dated 27 August 2011 or later approved revision.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>3046-CMS-01 Issue 1, Maintenance Manual Supplement 3046-CMS-01 Issue 1, Crew Manual Supplement 3046-SB-01 Issue 1, Service Bulletin later revisions of the above listed documents approved by EASA.</td>
<td>none</td>
<td>08/11/2010</td>
<td>Active</td>
<td>10032329</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Active</td>
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<td>10032331</td>
<td>JET AVIATION AG</td>
<td>REV. 2</td>
<td>747-400</td>
<td>Replacement of a dual Allied Signal High Frequency (HF) Radio by Dual Rockwell Collins Air Transport (ROAT) HF-8000D HF Radio Equipment (ARRINC735) with HF Data Link activation in accordance with JET Aviation Master Data List Doc. No. B7428343-008 Rev. D, dated 20 October 2010 or later approved revision.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List B7428343-008 Rev. D, dated 20 October 2010&lt;sup&gt;*&lt;/sup&gt; or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that there is no interaction between this design change and any other previously installed design change and/or repair will introduce a random effect upon the aerodynamic efficiency of the product.</td>
<td>29/11/2017</td>
<td>Active</td>
<td>10032331</td>
<td></td>
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<tr>
<td>10032336</td>
<td>S-TEC CORPORATION</td>
<td></td>
<td>FAA STC SA99320AC-D</td>
<td>CESSNA F182P, F182Q Installation of S-TEC System 20/30 Single and Two Axis Automatic Flight Guidance Systems, Model ST-762-2030 (14 Volt System)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificate/noise/approval or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA-approved S-TEC Bulletin No. 862, dated 29-May-09, Revision 6FAA-approved S-TEC Master Drawing List No. 921557, dated 10-Dec-09, Revision 3 for S-TEC System 20: FAA-approved Pilot's Operating Handbook and cockpit Flight Manual Supplement, PN B91921, Revision 3, dated 11-June-02 or later approved...</td>
<td>This STC is also eligible on Cessna models F182P and F182Q when modified by: 1. STC SA4658SW (Win conversion) and STC SA3825SW (260 H.P. engine conversion) or 2. STC SA4696CE (Rotax propellers) and/or STC SA285CE (Horton propellers) and STC SA3825SW (260 H.P. engine conversion) or 3. STCSA1382WE (Robert...</td>
<td>28/10/2010</td>
<td>Active</td>
<td>10032336</td>
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<tr>
<td>10032345</td>
<td>EICHELSDOERFER GmbH/FLUGZEUG BAU</td>
<td></td>
<td></td>
<td>RFS</td>
<td>STC Installation engine Rotor 912 and propeller HO-V53SF-51/S170SF in the powered saiplinary Fournier RF 5&lt;sup&gt;5&lt;/sup&gt;</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>STC &quot;Feinflugtechnische Mitteilung 002-09-01&quot; dated 27April 2010&lt;sup&gt;1&lt;/sup&gt;- Einbauanweisung 002-09-01, Einbau eines Rotax 912 A3 mit Propeller HO-V53SF-S170FG in den Motorsegler RF 5&lt;sup&gt;5&lt;/sup&gt; - Zeichnungsliste zum RFS-Rotax-Umbaus Flughandbuch für den Motorsegler Fournier RF 5, Rotor, HO-V532 vom 14.04.2010&lt;sup&gt;2&lt;/sup&gt;- Wartungsanleitung 27 April 2010&lt;sup&gt;2&lt;/sup&gt; Limited to Fournier RF 5, S/N 50/12</td>
<td>28/10/2010</td>
<td>Active</td>
<td>10032345</td>
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<tr>
<td>10032348</td>
<td>RAM AIRCRAFT, LP</td>
<td>REV. 1</td>
<td>FAA STC SA99671SC-D</td>
<td>2. CESSNA 414, 414A, 402C 1. CESSNA 340, 340A</td>
<td>Installation of Hartzell propeller PHC- C3YF-2UF/FC7693DF or PHC- C3YF-2UF/FC7693DF or PHC- C3YF-2UF/FC7693DF or PHC-</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>RMA Drawing No. 2549, Revision 2, dated May 2001&lt;sup&gt;3&lt;/sup&gt; - Aeronautical Flight Manual Supplement AFMS 1096, 2001&lt;sup&gt;3&lt;/sup&gt; - FAA STC SA99671SC-D, dated 10-December 2002&lt;sup&gt;4&lt;/sup&gt; - The Instruction of Continued Airworthiness are contained in Hartzell Propeller's Overhaul and Maintenance Manual No. 1170 (ATA 61-10-17), Revision 4, dated May 1999 or later.</td>
<td>1. Hartzell Model PHC- C3YF-2UF/FC7693DF or PHC- C3YF-2UF/FC7693DF or PHC- C3YF-2UF/FC7693DF or PHC- C3YF-2UF/FC7693DF or PHC-</td>
<td>02/11/2010</td>
<td>Active</td>
<td>10032348</td>
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<td>10032352</td>
<td>WIPARE Inc.</td>
<td>(EASA.IMA.S.01058)</td>
<td>US SAU91916Ch</td>
<td>AIR TRACTOR AT-802, AT-802A</td>
<td>Installation of vortex generators. Installation of Wipaire stall improvement array in accordance with Wipaire drawing No. 7A1-4407 Revision B, dated April 27,2010 for the AT-802A model and Installation drawing No. 1004245, Revision B, dated April 27,2010 for the AT-802 model</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAA STC No. SAU91916Ch amended May 24, 2010 &quot;Installation of Wipaire Stall Improvement Array&quot; for AT-802 and AT-802A models, FAA approved AFM for the AT-802 with PTBA-87F (1424nhp) engine and Wipaire stall improvement array dated May 10, 2010&lt;sup&gt;5&lt;/sup&gt; FAA approved AFMA...</td>
<td>FAA STC No. SAU91916Ch &quot;Limitations and conditions&quot;, amended May 24, 2010</td>
<td>05/11/2010</td>
<td>Active</td>
<td>10032352</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>Sicote Air FRANCE S.A.</td>
<td>A340-211, A340-212</td>
<td>AVIS Non IFE - Jordanian - New IFE and associated documentation update.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/ approval: CS-25 Amendment-8 Special Condition H-01: Enhanced Airworthiness Program for Aircraft Systems.</td>
<td>Change Approval Sheet. Ref: CS-A340-09-009</td>
<td>20/10/2010</td>
<td>Active</td>
<td>10032503</td>
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<tr>
<td>TENENCIA LTD.</td>
<td>A330-223</td>
<td>Installation of the Universal Aviation - Control of Compact DVD.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/ approval: EASA TCDS A-040CRN A-4001-002 EASA Supplemental Type Certification Basis.</td>
<td>Modification Approval Sheet. Ref: 5-05-2048</td>
<td>02/11/2010</td>
<td>Active</td>
<td>10032387</td>
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<td>Sabena Techronics BOD</td>
<td>AS355 F2, AS355 N, AS355 E, AS355 F1</td>
<td>Installation of the Intra Vist SRS 106-100 Cargo Hook</td>
<td>The Certification Basis (CB) for the original product remains applicable to the certificate/ approval.</td>
<td>AOC12 issued 2 issue 3 dated 28 August 2010.</td>
<td>None</td>
<td>26/04/2012</td>
<td>Active</td>
<td>10032391</td>
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<tr>
<td>AEROTECH &amp; CONCEPT</td>
<td>AS355 F2, AS355 N, AS355 E, AS355 F1</td>
<td>Installation of the Intra Vist SRS 106-100 Cargo Hook</td>
<td>The Certification Basis (CB) for the original product remains applicable to the certificate/ approval.</td>
<td>AOC12 issued 2 issue 3 dated 28 August 2010.</td>
<td>None</td>
<td>26/04/2012</td>
<td>Active</td>
<td>10032391</td>
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<td>Savannah Air Center, L.C.C.</td>
<td>FAA STC ST085B10</td>
<td>Installation, Complete Custom Interior, S/N 9260 Configuration</td>
<td>EASA TCDS No I.A.009 The Certification Basis for the original product remains applicable to the certificate/ approval.</td>
<td>FAA Oc. ref: ST0844AT. T dated 02 November 2010.</td>
<td>Restricted to AS9 9260 and 10.</td>
<td>04/11/2010</td>
<td>Active</td>
<td>10032416</td>
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<td>Teledyne Controls, LLC</td>
<td>FAA STC ST02340A.</td>
<td>Installation of the Wireless Ground Link</td>
<td>The Certification Basis for the original product remains applicable to the certificate/ approval.</td>
<td>Master Data List, Rev 2 dated 17 June 2010.</td>
<td>This STC is applicable to the following aircraft MSNs: 9260:</td>
<td>13/01/2022</td>
<td>Active</td>
<td>10032418</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>CAA UK RESPONSIBILITY</td>
<td>CAA UK RESPONSIBILITY</td>
<td>CAA UK RESPONSIBILITY</td>
<td>CAA UK RESPONSIBILITY</td>
<td>CAA UK RESPONSIBILITY</td>
<td>03/07/2022</td>
<td>Page 241 of 1,892</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC Ref.</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Movies</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10032443</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA7816SW-D</td>
<td>35-C3, F33C</td>
<td>Installation of S-TEC System 40/50 Single and Two Axis Automatic Flight Guidance Systems, Master ST-490-450 (28 Volt System)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unarchived and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with FAA Approved S-TEC Bulletin No. 590, dated 18-June-92. Approved S-TEC Master Drawing List No. 92596, dated 18-June-92, Operation in accordance with FAA Approved Airplane Flight Manual Supplement, P/N 89975, dated 26-June-92 is required for Beech models 35-C33 and F334...</td>
<td>none</td>
<td>08/11/2010</td>
<td>Active</td>
<td>10032443</td>
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<td>10032445</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>SA26-AT, SA226-AT, SA226-T, SA227-AT, SA227-AT, SA227-AT</td>
<td>Installation of Garmin GNS 530/540 as dual installation together with a Honeywell KWR 675 Marker Beacon Receiver</td>
<td>Design Non-significant change The Certification Basis of the original product and the following additional/alternative weight requirements are applicable to this change. For the change CS-23-23 has been elected to comply with where possible. Environmental Protection: The certificated noise and or...</td>
<td>EASA approved Engineering Order, Doc. No. 2081255-EO, issue 1, dated 26-Oct-2010, or later approved issue EASA approved Aircraft Flight Manual Supplement, Doc. No. 2081255-AMM, Issue 1, dated 25-Oct-2010(+), or later approved issue for GNS 43040/440 and GNS 530/540 EASA approved Aircraft Flight Manual Supplement</td>
<td>This change is limited to the configuration of aircraft / equipment types listed under Chapter 4 in EASA approved Engineering Order, Doc. No. 2081255-EO, issue 1, dated 28-Oct-2010, or later approved issue.</td>
<td>08/11/2010</td>
<td>Active</td>
<td>10032445</td>
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<td>10032446</td>
<td>AVIATION FABRICATORS, Inc</td>
<td>FAA STC SA2774CE</td>
<td>2, B300, B300C 2, B300C, B300CGT, B260CGT 1, B300CT, B260T, 300, 300LW 1, A200, A200C, A200CT, B260 1, 200, 200C, 200CT, 260T</td>
<td>Installation of Passenger Seats with Hinged Backrest</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and or emissions levels of the original product unarchived and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Aviation Fabricators Drawing List NumberAF-151 FAA Approved 12 November 1991 or later revisions of the abovementioned documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>none</td>
<td>08/11/2010</td>
<td>Active</td>
<td>10032446</td>
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<td>10032468</td>
<td>APICAL INDUSTRIES, Inc</td>
<td>REw. 1 FAA STC SW0220XLA</td>
<td>MBB-BK 117 C-2</td>
<td>Cable Cutters Kit Installation of Cable Cutters Kit with deflectors on skids and weper inaccordance with Master Document List MD 145-2, based on FAA STCS/R0220XLA</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>- Master Document List MD 145-2, Rev. A, dated August 16, 2010. Instructions for Continued Airworthiness ICA 145-21, Rev. NIC, dated September 23, 2012 or later revisions of the abovementioned documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>None</td>
<td>19/11/2010</td>
<td>Active</td>
<td>10032468</td>
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<tr>
<td>10032469</td>
<td>BUSINESS AND COMPUTER AIRCRAFT</td>
<td>C90</td>
<td>BCA18-00 - ADF system upgrade on beech S/N LJ - 894</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Limitation to beech C90 SN LJ-894</td>
<td>09/11/2010</td>
<td>Active</td>
<td>10032469</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>AIRQUIF A/S</td>
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<td></td>
<td>AS 350 B, AS 350 B4, AS 350 B1, AS 350 B2, AS 350 B3</td>
<td>Installation of Geophysical Equipment in accordance with Master Document List DO-350-25-107, revision 6, dated 09.11.2010</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement FMS-350-25-107, Revision 2, dated 19.08.2010</td>
<td>A cargo mirror allowing in-flight observation of the spectrometer and EMI magnetometer sensors must be installed</td>
<td>11/11/2010</td>
<td>Active</td>
<td>10032471</td>
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<tr>
<td>STANDARD AERO</td>
<td>REV. 1</td>
<td>FAA STC SA02367NY</td>
<td>F90 KING AIR</td>
<td>Installation of Pratt and Whitney Canada PT6A-135A Engines.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Modified with dual fuel feed lines, in accordance with Standard Aerocinc, Installation Instructions, Report, 55055-8, Rev. 0 dated 26 February 2007; - PT6A-135A engines must be modified in accordance with Supplemental Type Certificate SE02357NY prior to installation Note, if it is the installers responsibility</td>
<td>None</td>
<td>23/11/2010</td>
<td>Active</td>
<td>10032484</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 1</td>
<td>(EASA.R.S.01593)</td>
<td>AS 350 B/D/B1/B2/B4/B8/B3</td>
<td>Installation of Indraero Siren IS10-100 Cargo HookRevision 1: Transfer of EASA STC 10032496 from AERODAC to AEROTEC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>ADOC12 issue 2 dated 8 October 2010 - Certification Program AEC17 Issue 1 dated 8 October 2010 - Master Data ListADC54-27 Issue 1 dated 8 October 2010 - Instructions for Continued Airworthiness AD C04A issue 1 dated 18 September 2010</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/03/2016</td>
<td>Active</td>
<td>10032496</td>
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<td>ACS-NAI Ltd.</td>
<td></td>
<td>TCCA STC NO. SA10-25</td>
<td>DHC-8-401, DHC-8-402</td>
<td>Installation of LED Lavatory Light System JAR/CAS-25 as per EASA MPA. 191</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval plus: Special Condition H-01 &quot;Enhanced Airworthiness Programme for Aeronautic Systems - ICA on EWIS&quot; as detail...</td>
<td>ACC113 Instruction of FMS-350-25-107, Revision 1 dated 09.11.2010</td>
<td>None</td>
<td>12/11/2010</td>
<td>Active</td>
<td>10032519</td>
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<tr>
<td>VIKING AIR LIMITED</td>
<td></td>
<td>TCCA STC SA10-69</td>
<td>DHC-6 SERIES 400</td>
<td>Installation of RW Martin Air Conditioning System in accordance with TCCA STC SA10-69.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Flight Manual Supplement AC-S4-1A, Initial issue, TCCA approved September 5, 2016</td>
<td>None</td>
<td>12/11/2010</td>
<td>Active</td>
<td>10032527</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NA</td>
<td>Movie</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10032531</td>
<td>INNOMITECH AVIATION DIVISION OF IMP GROUP LTD</td>
<td>REV.1</td>
<td>TCCA STC NO. C-LSIA-10-007D</td>
<td>BD-700-1A10</td>
<td>Installation. Complete Custom Aircraft Interior - ASN 9374</td>
<td>EASA TCDS No. IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. Non-significant. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable...</td>
<td>TCCA Doc. C-10-0800 dated 12 January 2011. TCCA STC No. C-LSIA-10-02D Issue 1 dated 06 November 2010. AC/120200 Revision NA Flight Manual Supplement for Interior Completion. FMS-IA70-9374/Revision NC Flight Manual Supplement for CMC Electronic Flight Bag. FM1-10.</td>
<td>Restricted to AS 9374. Refer to the Limitations sections of the required AFM Supplement and/or the referenced FCOMS. 2.9. VLSI system is compatible only to IEEE 802.11b.</td>
<td>17/12/2011</td>
<td>Active</td>
<td>10032531</td>
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<td>10032532</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC NO. SAD4-113</td>
<td>BD-700-1A10</td>
<td>Installation. Complete Custom Aircraft Interior - ASN 9377</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/ approval. Non-significant. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this...</td>
<td>Transport Canada SoC reference C-10-0844 dated 10 November 2010. TCCA STC No. SAD4-113 Issue 1 dated 28 October 2004. AeriConsulting Services Configuration Definition List D85400 Revision FT dated 24 November 2010. Instructions for Continued Airworthiness - GO31770000-04A or later revisions of the above.</td>
<td>Restricted to AS 9377. Refer to the Limitations sections of the required AFM Supplement and/or the referenced FCOMS. 2.9. VLSI system is compatible only to IEEE 802.11b.</td>
<td>12/11/2010</td>
<td>Active</td>
<td>10032532</td>
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<td>10032536</td>
<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING 747-300 BOEING 747-200</td>
<td>Installation of a Dual Honeywell HT-9100 FMS per Modification Description TEN S39 Boeing 747-200 Dual Honeywell HT9100 GHSS Navigation Management System Installation, Issue 4 dated 10 November 2010</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>AFM supplement TEN SUPPLEMENT 1 ISSUE 3 for the Boeing 747 series Flight Manuals/Report TEN 539/7 Issue 2. Operations Manual Supplement Boeing 747/HT9100 FMS Installation. Later revisions of the above listed documents approved by EASA.</td>
<td>This STC is applicable to aircraft that have previously been modified in accordance with STC ST0181CH / M.L.A.S. 01660 GPS Antenna Provision Installation and have dual Honeywell Roll Computers P/N 2580263-914 or 915</td>
<td>12/11/2010</td>
<td>Active</td>
<td>10032536</td>
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<td>10032537</td>
<td>C&amp;D ASSOCIATES, Inc.</td>
<td>FAA STC SA02816CH</td>
<td>G68</td>
<td>Installation of C&lt;(&amp;&gt;)D Associates Combustion Heater PartNumber CD00599033.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Installation in accordance with C&lt;(&amp;&gt;)D Associates Installation Instructions/Revisions 539/10/00.</td>
<td>None</td>
<td>12/11/2010</td>
<td>Active</td>
<td>10032537</td>
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<td>10032538</td>
<td>AIRBUS HELICOPTERS TECHNICAL GMBH</td>
<td>REV.2</td>
<td>MBB-BK 117 C-2</td>
<td>MBB-BK 117 C-2</td>
<td>Main Gear Box Eurocopter BK 117 C2 Conversion of the Main Gear Box from MBB-BK117 C-2 Bluetooth standard PartNumber (PN) 17113240501 to MBB-BK117 C-2 Bluetooth standard PartNumber (PN) 1711324051 within the scope of a major overhaul, as documented in the ZFL Luftfahrttech ZFL major change STC03, in accordance with...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Airworthiness Limitations - ZFL doc. no. STC03-1682/1109. &quot;Supplement for ECD Master Servicing Manual MBB-BK117 C-2, Chapter 24-00-00, 2-1&quot;, Issue 2r later revisions of the above listed documents approved by EASA.</td>
<td>- This conversion can only be performed by the STC holder. After conversion to Main Gear Box PN 1711324051-001 this item is to be installed on MBB-BK117 C-2 helicopters.</td>
<td>13/06/2012</td>
<td>Active</td>
<td>10032538</td>
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<td>EASA STC No</td>
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<td>Rev. level</td>
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<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10032602</td>
<td>AVIN CIS AVIATION ITALIA S.P.A</td>
<td>REV. 3</td>
<td>650</td>
<td>Enhanced Ground Proximity Warning System (EGPWS) Honeywell MK VIII installation in replacement of Honeywell GWPS MXV Rev. 2 is an administrative re-issuance to reflect the recent namechange from INAER Aviation Italia S.p.A. Divisione Ala Fissa Rev. 3: Transfer of...</td>
<td>Master Drawing List MDX-34-44-1003-A, Revision 00 from 10 Nov. 2010- Aircraft Flight Manual Supplement, FMS-34-44-1003-A, Revision 00 from 18May 2010-Instructions for Continued Airworthiness ICA-34-44-1003-A, Revision 00from 04 June 2010-SAN statement of compliance to DOA SAN project No. 434-1003-A...</td>
<td>STC is applicable to S/N 712 and Up&lt;(&gt;,&lt;)&gt;3. Placard 13A690-1 stating &quot;Conduct trim check prior to first flight after (see PDRI) is required. 1. This STC is limited to aircraft Models PA-31, PA-31-325 and PA-31-350&lt;(&gt;,&lt;)&gt;2. Piper Model PA-31 is limited to S/N 712 and Up&lt;(&gt;,&lt;)&gt;3. Place...</td>
<td>22/02/2024</td>
<td>Active</td>
<td>10032602</td>
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<td>10032603</td>
<td>AIR METHODS CORPORATION</td>
<td>REV. 3</td>
<td>429</td>
<td>Installation of a controllable Searchlight</td>
<td>Master Drawing List MDX-34-44-1003-A, Revision 00 from 10 Nov. 2010- Aircraft Flight Manual Supplement, FMS-34-44-1003-A, Revision 00 from 18May 2010-Instructions for Continued Airworthiness ICA-34-44-1003-A, Revision 00from 04 June 2010-SAN statement of compliance to DOA SAN project No. 434-1003-A...</td>
<td>STC is applicable to S/N 712 and Up&lt;(&gt;,&lt;)&gt;3. Placard 13A690-1 stating &quot;Conduct trim check prior to first flight after (see PDRI) is required. 1. This STC is limited to aircraft Models PA-31, PA-31-325 and PA-31-350&lt;(&gt;,&lt;)&gt;2. Piper Model PA-31 is limited to S/N 712 and Up&lt;(&gt;,&lt;)&gt;3. Place...</td>
<td>17/11/2010</td>
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<td>10032603</td>
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<td>10032604</td>
<td>TDQ AEROSPACE INC</td>
<td>REV. 3</td>
<td>8767-200, -300, -300F</td>
<td>Installation of Universal Fault Interrupter (UFI) for Centre Wing FuelTank Electrical Fault Protection.</td>
<td>Master Drawing List MDX-34-44-1003-A, Revision 00 from 10 Nov. 2010- Aircraft Flight Manual Supplement, FMS-34-44-1003-A, Revision 00 from 18May 2010-Instructions for Continued Airworthiness ICA-34-44-1003-A, Revision 00from 04 June 2010-SAN statement of compliance to DOA SAN project No. 434-1003-A...</td>
<td>STC is applicable to S/N 712 and Up&lt;(&gt;,&lt;)&gt;3. Placard 13A690-1 stating &quot;Conduct trim check prior to first flight after (see PDRI) is required. 1. This STC is limited to aircraft Models PA-31, PA-31-325 and PA-31-350&lt;(&gt;,&lt;)&gt;2. Piper Model PA-31 is limited to S/N 712 and Up&lt;(&gt;,&lt;)&gt;3. Place...</td>
<td>17/11/2010</td>
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<td>EASA STC no.</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10032609</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC STD42040AT-D</td>
<td>GVX-X</td>
<td>Aircraft Completion and Installation of an Airplane Interior</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certifications of noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>You may not use descriptive data about this design change to duplicate other products. This approval is limited to aircraft Serial 4200. This STC does not permit manufacturing of parts for multiple installations.</td>
<td>17/11/2010</td>
<td>Active</td>
<td>10032609</td>
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<td>10032617</td>
<td>AIR METHODS CORPORATION</td>
<td>FAA STC SR00714DE-D</td>
<td>BELL 429</td>
<td>Installation of Emergency Medical Services Interior and Systems</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certifications of noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>17/11/2010</td>
<td>Active</td>
<td>10032617</td>
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<tr>
<td>10032620</td>
<td>AIR METHODS CORPORATION</td>
<td>FAA STC SR00710DE-D</td>
<td>BELL 429</td>
<td>Installation of Scene and Patient Loading Lights</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certifications of noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>17/11/2010</td>
<td>Active</td>
<td>10032620</td>
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<td>10032621</td>
<td>AIR METHODS CORPORATION</td>
<td>FAA STC SR00698DE-D</td>
<td>BELL 429</td>
<td>Installation of Patient Loading System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certifications of noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>17/11/2010</td>
<td>Active</td>
<td>10032621</td>
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<td>10032628</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A319, A320, A321</td>
<td>New European Cabin Modification</td>
<td>Certification Basis in accordance with EASA Type Certificate Data Sheets A.064. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable: - Supplemental Type Definition Document, ref. doc. 252051-001M, Issue 1, dated 17th November 2010 - Compliance Record List, ref. doc. CRB25-101M, Issue 1, dated 17th November 2010 - Applicant's Declaration of Compliance No. 2020, dated 18th November 2010 for later revisions of the above listed document...</td>
<td>Not applicable</td>
<td>18/11/2010</td>
<td>Active</td>
<td>10032628</td>
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<td>STC no.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STD</td>
<td>ST04213AT-D</td>
<td>GIV-X</td>
<td>Aircraft Completion and Installation of Airplane Interior.</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index GC144914D00, Revision J, dated November 06, 2010, for Continued Airworthiness.</td>
<td>You cannot use descriptive data about this design change to duplicate other products. This approval is limited to Gulfstream GIV-X, Serial 4202. This STC does not permit manufacturing of parts for multiple installations.</td>
<td>18/11/2010</td>
<td>Active</td>
<td>10032631</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC</td>
<td>ST03742AT-D</td>
<td>GIV-X</td>
<td>Installation of a Cabin/Baggage Compartment.</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List GC420004171, Revision G, dated November 02, 2010, for Operation in accordance with Airplane Flight Manual Supplement GC420041M009, Revision NC, dated November 20, 2009, or later revisions of the above listed documents approved.</td>
<td>None</td>
<td>19/11/2010</td>
<td>Active</td>
<td>10032651</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>A340-600</td>
<td>A340-600 Cabin Retrofit - ETD05</td>
<td>Certification Basis in accordance with EASA Type Certificate Data Sheet No. 015. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>CRI H-01: “Enhanced Airworthiness Programme for Aeroplanes...” - Supplemental Type Definition Document, ref. doc. STDD00-10/23M, Issue 1, dated 23rd November 2010, Compliance Record List, ref. CRL00-10/23M, Issue 1, dated 23rd November 2010, Applicant’s Declaration of Compliance No. 2691, dated 23rd November 2010, or later revisions of the above listed document.</td>
<td>Not applicable</td>
<td>23/11/2010</td>
<td>Active</td>
<td>10032691</td>
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<td>ACS-NAI Ltd. (EASA:MA.S.0310)</td>
<td></td>
<td>REV. 2</td>
<td>TCCA LSTC C-L4A09-024/D, CL-600-2B19</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA LSTC No. C-L4A09-024D Issue 3 dated 26 October 2010, AERO Consulting Services Ltd. Modification Data Summary G102000 Revision 0, Midcoast Aviation Inc. Aircraft Flight Manual Supplement for the Complete Custom Aircraft Interior Installation, S3-258M008 Rev NC, AERO Consulting Services Ltd.</td>
<td>Restricted to ASN 8077 Prior to installation of this design change it must be determined the relationship between this design change and any other previously installed design change and/or repair will introduce inadverse effect upon the airworthiness of the product.</td>
<td>16/06/2013</td>
<td>Active</td>
<td>10032736</td>
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<td>EASA STC ref.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10032737</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>REV. 1</td>
<td>FAA STC SR08M05E</td>
<td>407, 206L, 206L-1, 206L-3, 206L-4</td>
<td>Cargo Hook Kit model 200-260-00 and 200-260-01 This change adds an additional cargo hook</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EASA TCDS IM.A.033 The Certification Plan 033-02-A-2103-R00 dated 25/11/2010 Certification Plan 033-02-A-2101 rev 01</td>
<td>Applies only to those Bell Helicopter models listed above previously equipped with Bell Helicopter Cargo Hook Suspension System DR-706-314-109</td>
<td>26/11/2010</td>
<td>Active</td>
<td>10032737</td>
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<tr>
<td>10032763</td>
<td>SABENA TECHNICS BOC</td>
<td></td>
<td></td>
<td></td>
<td>Onboard Mobile Phone System installation on Boeing 777-300ER aircraft provisioned by Boeing</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The Certification Plan 033-02-A-2103-R00 dated 25/11/2010 Certification Plan 033-02-A-2101 rev 01</td>
<td>None</td>
<td>None</td>
<td>30/11/2010</td>
<td>Active</td>
<td>10032763</td>
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<tr>
<td>10032771</td>
<td>LIFEPORT, LLC</td>
<td>FAA STC ST02296LA</td>
<td>CESSNA 680</td>
<td>Installation of a Medical Patient Loading and Utility System (PLUS)</td>
<td>EASA TCDS IM.A.033 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List MDL-845-00, Revision T from 11 Nov. 2009 - Airplane Flight Manual Supplement EGR-845-11, Revision A from 22 Feb. 2010</td>
<td>None</td>
<td>None</td>
<td>30/11/2010</td>
<td>Active</td>
<td>10032771</td>
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<tr>
<td>10032781</td>
<td>SOCIETE D’ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td></td>
<td>DR400/NGL, DR400/2-2, DR400/500, DR400/180, DR400/180 5, DR400/200 R, DR400/180, DR400/180 R, DR400/160, DR400/160 D, DR400/140, DR400/140 B, DR400/120, DR400/120 A, DR400/120 D, DR400/125, DR360, DR350, DR400/100, DR400/180 R, DR400/100, DR350, CR340, DR400/125, DR300/125, DR300/140, DR300/120, DR300/108, DR300/120, CR340, DR300/108, DR300/120, CR340, DR300/108, DR300/120</td>
<td>A-07-09-Filtress Ed. 0 - Installation of remote oil filter kit</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The Certification Plan 033-02-A-2103-R00 dated 25/11/2010 Certification Plan 033-02-A-2101 rev 01</td>
<td>Document of Application A-07-09-Filtress DA du 15 November 2010</td>
<td>None</td>
<td>None</td>
<td>01/12/2010</td>
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<td>STC Holder</td>
<td>FAA STC No.</td>
<td>Associated Technical Documentation</td>
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<td>Limitations/Conditions</td>
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<td>System Record</td>
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<tr>
<td>AAKCO TECHNICAL SERVICES INC.</td>
<td>10032878</td>
<td>FAA ST0574AT</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition: H-01 ICA on EWS166e following paragraph(s) at a later amendment: affected paragraphs at 352 amnds. 4, 25.601, 25.603, 25.605, 25.606, 25.613, 25.1203,...</td>
<td>Made with Data Util ref MDL-001, Revision 7Airbus A320 / A310 Installation Instructions A0-001 Revision 2, dated 01 October 2010Aibus A318 / A319 / A320 / A321 Installation Instructions A2-001Revision 2, dated 01 October 2010Airbus A330 / A340 Installation Instructions A3-001 Revision 2, dated 01 October 2010</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>02/06/2021</td>
<td>Active</td>
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<td>GAMA ENGINEERING Ltd.</td>
<td>10032795</td>
<td>CAA UK RESPONSIBILITY</td>
<td>114, 114A, 114B, 114TC, 112TC, 112TC,</td>
<td>Aspen EF1000 Software Version 2.3.1 upgrade</td>
<td>Flight Manual Supplement AAMS-1108, Issue 2Installation Instructions for Continued Airworthiness ICA-1108, Issue 2or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>03/05/2012</td>
<td>Active</td>
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<tr>
<td>VAN REEMSDUIK ROTTERDAM B.V.</td>
<td>10032842</td>
<td>REV. 15</td>
<td>777F 747-200F, 747-400F, 747-8F 2 777F 1 747-200F-400F-8F</td>
<td>RGX-0517 Special Cargo ContainerCargo container with a pallet size of 236.5 x 96 with P/N 136.080-0044 P/N 136.080-080, P/N 136.080-100, P/N 136.080-110 and P/N 136.080-044C P/N 136.080-086 Rev 1: typo corrected in the limitations section Rev 2: 8774-8F and 777F added to the appl...</td>
<td>The Type and OTOF Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated airworthiness noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The loading limitations from the RGX container are according to the limitations in the applicable WIN&lt;-&gt;I&lt;-&gt;P manuals for the NAS2610 aiccode G ULD's. The installation of this modification by third persons is subject to written permission by the approval holder. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>27/06/2024</td>
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<tr>
<td>Rotorcraft Support, Inc.</td>
<td>10032879</td>
<td>FAA STC</td>
<td>407</td>
<td>Installation of Concorde RG407 Battery</td>
<td>Flight Manual Supplement AAMS-1108, Issue 2Installation Instructions for Continued Airworthiness ICA-1108, Issue 2or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>06/12/2010</td>
<td>Active</td>
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<tr>
<td>Luftansa Technik AG</td>
<td>10032887</td>
<td>B737-500</td>
<td>DLH New European Cabin for 737-500 Aircraft. Modification of the Aircraft seat layout to increase the seat capacity from currently 111 seats to 120 new seats, produced by Recaro/Aircraftseating, Germany.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document Ref. STD001-100M Issue 01 dated 01 December 2010 Compliance Record Document Ref.CRL00-10/03M dated 01 December 2010. Applicant's Declaration of Compliance No.2770 dated 02 December 2010 or later revisions of the above listed documents approved by EASA.</td>
<td>Not Applicable</td>
<td>06/12/2010</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10032893</td>
<td>INNOVATIVE SOLUTIONS AND SUPPORT, Inc.</td>
<td>FAA STC SA02255NY</td>
<td>PC-12</td>
<td>PC-12 34-66-802755 FPD/FAA SA02255NYConfiguration A - Dual 15&quot; Flat Panel Display System (Plat and Copilot Positions) Installation of Innovative Solutions &amp; Support - (5)(6)/(7)/(8)Dual 15&quot; Flat Panel Display System in accordance with IS&gt;(&gt;/&amp;&lt;)&gt;S Master Drawings, ID-0919 Revision 9, dated March 27...,</td>
<td>FAR 23 The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval</td>
<td>Configuration A - Dual 15&quot; Flat Panel Display System (Plat and Copilot Positions) IS&gt;(&gt;/&amp;&lt;)&gt;S Airplane Flight Manual Supplement Document No. 10-08314, revision C, FAA Approved April 7, 2009 is required with the installation Configuration B - 10&quot; Flat Panel Display System (Plat and Copilot Positions)</td>
<td>Reg.: 1/- S/N: PC-12, /46, /47</td>
<td>08/12/2010</td>
<td>Active</td>
<td>10032893</td>
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</tr>
<tr>
<td>10032896</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC 7010HSE.D</td>
<td>B737-500</td>
<td>Interior Reconfiguration in accordance with Flight structures, Inc. (FSI) Master Drawing List B0F5003-001 Installation of honeycomb, slat/strake, slat/costal extension kits, movable under-lin class divider, side wall panel, lowered ceiling panel and associated installation hardware.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. FAR 23, Amendment 1-23 The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Flight structures, Inc (FSI) Master Drawing List B0F5003-001 Rev.Cdated 01-Dec-2010. Certification Plan Summary B0F5003-D10-1 Rev.C dated 02 December 2010. Supplement to Instructions for Continued Airworthiness (ICA) B0F5003-D15 rev.B dated 19 November 2010 or later revisions of the above listed documents approved by EASA</td>
<td>Not Applicable</td>
<td>07/12/2010</td>
<td>Active</td>
<td>10032896</td>
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<tr>
<td>10032902</td>
<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td>SC 01, SC 01B</td>
<td>Inspection Program to increase operating hours Implementation of Inspection Program to increase operating hours</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. FAR 23, Amendment 1-23 The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Maintenance Manual Supplement MM-100500000, Rev. 1or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>08/02/2021</td>
<td>Active</td>
<td>10032902</td>
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<tr>
<td>10032903</td>
<td>AVIATION FABRICATORS, Inc.</td>
<td>FAA STC REF 5A071/5SW</td>
<td>SEE EASA APPROVED MODELLIST</td>
<td>Installation of one, two, or three place side facing seat (frame only).</td>
<td>FAR 23, plus Amendments listed in the EASA or FAA TCDS, as appropriate to the Model The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Aviation Fabricators Drawing ref B6912 dated 7/4/89, or drawing D-28R813143. The aircraft must be equipped with a third radionavigation oridocommunicator mean independent from the two GNS 430 units. The installation of a further version must be approved</td>
<td>None</td>
<td>07/12/2010</td>
<td>Active</td>
<td>10032903</td>
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<tr>
<td>10032904</td>
<td>ASI INNOVATION</td>
<td>F406</td>
<td>ASI-FAM-0015-02 Provision installation of console PAN3Q69 and LH/Whitney</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>Limited to: F406 S/N 006-017-025-039-042-070-074-075-077-086</td>
<td>07/12/2010</td>
<td>Active</td>
<td>10032904</td>
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<td>10032905</td>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td>PA-28RT, PA-201T,</td>
<td>Avionic modification: dual installation of Garmin GNS 430W (with GA 31antennas) for B-RNAV operation with coupling to autopilot, installation of GMA 340 audio Intercom, KN 62A DME receiver and WX-500 storm/ice warning NY183 antenna</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>SEAM Flight Manual Supplement No 3049/10 Revision 1, dated 22-October 2010SEAM Modification Fix No 3049/10 Revision 1, dated 22-October 2010or later revisions of the above listed documents approved by EASA</td>
<td>The installation is limited to aircraft PA-28RT/PA-28T S/N 269813143. The aircraft must be equipped with a third radionavigation oridocommunicator mean independent from the two GNS 430 units. The GNS 430W software version must be No 3.3. Installation of a further version must be approved</td>
<td>07/12/2010</td>
<td>Active</td>
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<td>STC No.</td>
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<td>STD No.</td>
<td>STC Type</td>
<td>Description</td>
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<td>10032907</td>
<td>BOMBARDIER INC.</td>
<td>STC 2010-006</td>
<td>TCCA STC NO. SA04-113</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9367</td>
<td>EASA.M.A.009 The Certification Basis for the original product remains applicable to this modification: approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Transport Canada SoC reference C-10-0043 dated 02 December 2010 TCCA STC No. SA04-113 issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision FW dated 26 November 2010. Restrictions for Continued Airworthiness GC93710000-FC/AIR later revisions of the above ... Restricted to ASN 9367 and/or to the limitations sections of the required AFM Supplement and/or to the referenced FCMS2.2.WLAN system is compatible only to IEEE 802.11(b).3 Prior to installation of this modification the installer must determine that the interrelationship between this modification and ...</td>
<td>07/12/2010</td>
<td>Active</td>
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<td>10032919</td>
<td>BOMBARDIER INC.</td>
<td>STC 2010-006</td>
<td>TCCA STC NO. SA04-113</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9371</td>
<td>EASA.M.A.009 The Certification Basis for the original product remains applicable to this modification: approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Transport Canada SoC reference C-10-0041 dated 25 November 2010 TCCA STC No. SA04-113 issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List D954000 Revision PW dated 19 November 2010. Restrictions for Continued Airworthiness GC93710000-FC/AIR later revisions of the above ... Restricted to ASN 9371 and/or to the limitations sections of the required AFM Supplement and/or to the referenced FCMS2.2.WLAN system is compatible only to IEEE 802.11(b).3 Prior to installation of this modification the installer must determine that the interrelationship between this modification and ...</td>
<td>08/12/2010</td>
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<td>10032965</td>
<td>FIELD AVIATION COMPANY INC.</td>
<td>STC 2010-006</td>
<td>TCCA STC NO. SA04-113</td>
<td>DHC-8-311</td>
<td>Installation of AR Baggage Bay Divider Nets TCCA STC CL-SA60-212/D</td>
<td>FAR 25 as per EASA M.A. 191 The Certification Basis for the original product remains applicable to this modification: approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. FAA STC ST020738E dated July 26, 2004 IAE Certification Plan. LAP1009R10. Revision A dated March 11, 2010. FAA Master Drawing List. LAP 1029-001. Revision A dated June 23, 2010. Installation Drawing LAP 2009. Revision B dated June 23, 2010. Restrictions for Continued Airworthiness GC93710000-FC/AIR later revisions of the above ... The modification is applicable only when the necessary structural provisions are pre-existing.</td>
<td>10/12/2010</td>
<td>Active</td>
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<td>10032968</td>
<td>BAYO S.A.S.</td>
<td>STC 2010-006</td>
<td>HR 100285 T14RA</td>
<td>STC 07-006</td>
<td>Installation of single EFIS Aspen EFD1000 Pro for VFR navigation.</td>
<td>The Certification Basis for the original product remains applicable to this modification: approval. The certificate noise or emissions levels of the original product unchanged and remain applicable to this certificate/approval. Bayo AFM supplement n ° 2010-006, revision 3 dated 09/12/2008 Bayo Master list no. 2010-006-ML, revision 4 dated 09/12/2010 Bayo modification file n° 2010-006-DEF, revision 2 dated 30/05/2010. Approved software version for ACU and PFD units is 1.1 Navigation coupling to automatic pilot is limited to heading mode.</td>
<td>10/12/2010</td>
<td>Active</td>
<td>10032968</td>
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<tr>
<td>10032969</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>STC 2010-006</td>
<td>CAA UK RESPONSIBILITY</td>
<td>REV. 2</td>
<td>Activation of Nordicam WLAN System</td>
<td>The Certification Basis for the original product remains applicable to this modification: approval. The requirements for environmental protection and the associated certificate noise or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. H4 Aerospace Modification Summary No. MOD H4AD13434 issue 4 dated 05 May 2011. H4MS No. H4MS003H4CA No. H4ADRS60 Issue 20r late revisions of the above listed documents approved by IAEASA. Restricted to ASN 9011 Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/06/2016</td>
<td>Active</td>
<td>10032969</td>
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### List of EASA Supplemental Type Certificates

<table>
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<tr>
<th>STC Holder</th>
<th>FAA STC ST-0215AT-D</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>Gulfstream Aerospace Corporation</td>
<td>GIV-X Aircraft Completion and Installation of an Airplane Interior on Aircraft Serial 4203.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC414914203, Revision F, dated November 09, 2010. Aircraft to be maintained in accordance with Instructions for Continuation/Earmark Deviations Supplement GC425014431, Revision A, dated November 09, 2010, or later revisions.</td>
<td>None</td>
<td>13/12/2010</td>
<td>Active</td>
<td>10032980</td>
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<tr>
<td>Hawker Beechcraft Ltd.</td>
<td>B200, B300 Installation of Thrane &amp; Thrane Aviator 200/300 Inmarsat Satcom with/Wi-Fi capability on Beechcraft King Air B200/B300</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installed to Major Change SC0353A at issue 1. Operation in accordance with Hawker Beechcraft Limited Airplane Flight Manual Supplement X-King Air B200: 101-590168 Supplement 1, Issue X-King Air B300: 130-590031 Supplement 1, Issue 1. Aircraft to be maintained in accordance with Instructions for Continuation/Earmark Deviations Supplement GC425014431, Revision A, dated November 09, 2010, or later revisions.</td>
<td>This modification is only applicable to aircraft equipped with the Collins Proline 21 avionics package. Refer to aircraft defined in Hawker Beechcraft major change SC0353A.</td>
<td>13/12/2010</td>
<td>Active</td>
<td>10032984</td>
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<tr>
<td>Aviation Service A.S.</td>
<td>AN-26-B An-26 Aircraft Upgrade: TCAS II CAS-67A, Mode S Transponder, MST-67A and/773, EGPWS Mk-XV, NAV/COM MSK165A, GP155XL, Audiolap panel KM429, VHF/FARST 2000, MFD KMD540, ADF K987, DME K683, Air Data Display ADII.08.RR 904458, Airspeed Ind. LUN110, ELC G468-D</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CRI A-100: The requirements for environmental protection and the associated certificated noise and or emissions levels of the original product...</td>
<td>- Modification Approval Sheet, ref. Modification S-46-2102, Issue 2: dated 10th December 2010; Certification Program, ref. DO/N/2009/226/DZ, Issue 3, dated 10th December 2010; Airplane Flight Manual Supplement, ref. AFM 5-D032-01, Rev. C. Later revisions of the above listed documents approved by EASA.</td>
<td>Not applicable</td>
<td>15/12/2010</td>
<td>Active</td>
<td>10033002</td>
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<td>Sabena Technics BOD</td>
<td>A321 All Models A330 All Models A319 All Models A318 All Models Full provision for EFB Class 2 on Single Axis Aircraft</td>
<td>Certification Basis in accordance with EASA Type Certificate Data Sheet A.004: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CRI H-01 “Enhanced Airworthiness Programme for Aircraft...”</td>
<td>- Modification Approval Sheet, ref. Modification S-46-2102, Issue 2: dated 10th December 2010; Certification Program, ref. DO/N/2009/226/DZ, Issue 3, dated 10th December 2010; Airplane Flight Manual Supplement, ref. AFM 5-D032-01, Rev. C. Later revisions of the above listed documents approved by EASA.</td>
<td>Not applicable</td>
<td>15/12/2010</td>
<td>Active</td>
<td>10033002</td>
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<tr>
<td>B/E Aerospace, Inc. - FSI</td>
<td>B737-300 Interior Reconfiguration in accordance with Flight Structure, Inc. (FSI) Master Drawing List B0FS003-D02: Installation of new windshield, storage bin, storage bin extension kits, movable under-lam class divider, sidewall panel, lowered ceiling panels, and associated installation hardware.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- FSI Master Drawing List B0FS003-D02 Rev C: dated 11-Dec-2010; Certification Program Summary B0FS003-D10-C: dated 13-Dec-2010; Supplement to Instructions for Continued Airworthiness (ICA) B0FS003-D15 Rev. C: dated 19 November 2010; FAA “Other revisions of the above listed documents approved by FAA...”</td>
<td>None</td>
<td>15/12/2010</td>
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<tr>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>TBM 700 N TBM 700 C1, TBM 700 C2, TBM 700 A, TBM 700 B, Installation of 5-blade constant speed propeller,MTV-27-1,E-C-F-R/PY CFR2065-FD propeller on TBM 700 A, TBM 700 B, TBM1, C1, TBM 700 N (tradename TBM850), TBM 700 N (tradename TBM900)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>According to MT-Propeller Master Document List No. 02 2011 Rev.17o be operated in accordance with MT-Propeller AFM-S Document No. E-1870 or later revisions. The above listed documents approved by EASA. To be installed as: Installation Instructions Doc. No. E-1870, or later.</td>
<td>According to MT-Propeller Master Document List No. 02 2011 Rev.17o to be operated in accordance with MT-Propeller AFM-S Document No. E-1870 or later revisions. The above listed documents approved by EASA. To be installed as: Installation Instructions Doc. No. E-1870, or later.</td>
<td>07/02/2015</td>
<td>Active</td>
<td>10033038</td>
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<td>PETERSEN AVIATION, INC.</td>
<td>FAA STC SA2670CE</td>
<td>Operation of airplanes using automotive gasoline according US STC SA2670CE</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
<td>- According to Amdt 19.- CRI amendment: CS 25, App.S25.50 following paragraph(s) at a later date.</td>
<td>17/12/2010</td>
<td>Active</td>
<td>10033096</td>
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<tr>
<td>AEROLITE AG</td>
<td>EC135</td>
<td>VIP Equipment P/N 135748-501; JAR 27 Issue 1 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<td>17/12/2010</td>
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<td>10033098</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>A340-313 A340-300</td>
<td>A340-300: “GAP” VIP Cabin Interior, CompletionRev. 1 introduces a reduction of the Minimum Cabin Crew on a number of fl.</td>
<td>The Type and OOS Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25, App.S25.50 (a)as Amtd 19.; - CRI A-2301-002EASA Supplemental Type Certification Basis for Lufthansa Tech.</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between the following change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>11/11/2021</td>
<td>Active</td>
<td>10033112</td>
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(EASA.A.S.03762) | A340-300 | Installation of VIP Cabin Interior, CompletionRev. 1 introduces a reduction of the Minimum Cabin Crew on a number of fl. | The Type and OOS Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25, App.S25.50 (a)as Amtd 19.; - CRI A-2301-002EASA Supplemental Type Certification Basis for Lufthansa Tech. | | Prior to installation of this change/repair it must be determined that the interrelationship between the following change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft. | 11/11/2021 | Active | 10033112 |

<p>| AVIODYNE CORPORATION | FAA STC SA00298BO | SR20, SR22 | Installation of: Aviodyne Model DFC00 Autopilot P/N 700-00170-000 System, loaded with Release 2 software. | Aviodyne Model DFC100 Autopilot P/N 700-00170-001 System, loaded with Release 2 software. | As indicated in FAA STC SA00298BO aircraft must be configured with an existing STC: Aviodyne Corporation Autopilot, Existing S-Tec, S-Tec Part Number 30ALT, 01260/-/... | 06/03/2014 | Active | 10033140 |</p>
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<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>ROEDER PRAEJSION GMBH</td>
<td>REV. 1</td>
<td>SA26-T, SA26-AT, SA26- T</td>
<td>Dual Installation of GNS 430 WAAS in SA26/227 Series</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the STC certificate only.</td>
<td>GNS430-NV-03157-05, Rev. 2, dated 05.08.2010</td>
<td>- Flight Manual Supplement: GNS430-NV-03157-05, Rev. 3, dated 05.08.2010</td>
<td>Installation of 3rd dissimilar and independent COM system required to/for operation.</td>
<td>15/11/2017</td>
<td>Active</td>
<td>T303175</td>
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</table>

| AVIONITEC AG | REV. 1 | T303 | Avionics Retrofit/Removal - Garmin GNS 430 NAV/COM/GPS/ King XX 165 NAV/COM with King N 204 Indicating ARC RTA 47A DME-ARC R 446A ADF with ARC IN 36A Indicator/detachable - Garmin GNS330NAV/COM/GPS (upgrade from a Garmin GNS 430) - Garmin GNS330 NAV/COM/GPS with Garmin GI 106A Ind.-King IN63A DME | The Certification Basis for the original product remains applicable to the STC certificate only. The certificated hardware emissions levels of the original product are unchanged and remain applicable to this certification. | Document No. P161455-017, Rev. 0-3 | FAA STC ST01827LA must be installed in conjunction with this STC | | 1/12/2010 | Active | T303155       |

| GARMIN INTERNATIONAL, Inc. | REV. 9 | SA015359W1-D | Installation of the Garmin G1000 Integrated Avionics System and optional Electronic Stability and Protection (ESP). Rev. 7 adds two new models and updates the Master Data List (MDL) for the B260 and B300 models. Rev. 8 is a correction and re-introduces the merit. | The Certification Basis for the original product remains applicable to the STC certificate only. White areas are where amended by additional or later amendments if indicated on FAA STC SA015359W1-D. The requirements for environmental protection and the associated identified noise or emissions levels of the product are unchanged. | FAA STC ST01827LA must be installed in conjunction with this STC | 09/06/2022 | Active | T303158       |

| ANDRA INTERNATIONAL LLC | REV. 3 | FAA STC ST02160A | Installation of main deck cargo loading system | The Certification Basis for the original product remains applicable to the STC certificate only. | FAA STC ST01827LA must be installed in conjunction with this STC | 21/12/2010 | Active | T303161       |

| NSE | REV. 6 | EC135 T3 (CPD) | Installation of a Night Vision Imaging System (NVIS) enabling operation. | The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements are applicable to the STC certificate only. | - Rotorcraft Flight Manual Supplement RPM-EC135-NVIS Revised 06, including: Appendix A: RPM (including NVIS kit configuration) | | 2709/2023 | Active | T303164       |

| INNOVATIVE SOLUTIONSAND SUPPORT, Inc. | REV. 3 | FAA ST03730C | Installation of an Innovative Solutionsand Support (ISAS) Flight Safety System (FSS) and of an L-3 Avionics systems GIS-3100 Electronic Standby Instrument System (ESIS). Rev. 6 rephrases the paragraph on limitations. Rev. 3 corrects a typographical error. Rev. 3 adds the model 757-7. | The Certification Basis for the original product remains applicable to the STC certificate only. | FAA STC ST03730C must be installed in conjunction with this STC | 22/03/2021 | Active | T303170       |

| EASA (UENA, 2004/04/01) | REV. 1 | FAA ST03730C | Installation of a Night Vision Imaging System (NVIS) enabling operation. | The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements are applicable to the STC certificate only. | - Rotorcraft Flight Manual Supplement RPM-EC135-NVIS Revised 06, including: Appendix A: RPM (including NVIS kit configuration) | | 2709/2023 | Active | T303164       |

| EASA (UENA, 2004/04/01) | REV. 1 | FAA ST03730C | Installation of a Night Vision Imaging System (NVIS) enabling operation. | The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements are applicable to the STC certificate only. | - Rotorcraft Flight Manual Supplement RPM-EC135-NVIS Revised 06, including: Appendix A: RPM (including NVIS kit configuration) | | 2709/2023 | Active | T303164       |

<p>| EASA (UENA, 2004/04/01) | REV. 1 | FAA ST03730C | Installation of a Night Vision Imaging System (NVIS) enabling operation. | The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements are applicable to the STC certificate only. | - Rotorcraft Flight Manual Supplement RPM-EC135-NVIS Revised 06, including: Appendix A: RPM (including NVIS kit configuration) | | 2709/2023 | Active | T303164       |</p>
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<th>Row</th>
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<th>Rev. level</th>
<th>Foreign/NAA Aircraft Cert.</th>
<th>Movie</th>
<th>Description</th>
<th>Certification Basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10033177</td>
<td>BOBBARDIER INC.</td>
<td>-</td>
<td>TCCA STC NO. SA94-113</td>
<td>8D-780-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9369</td>
<td>EASA A.M.2097: The Certification Basis for the original product remains applicable to this certificated approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>Transport Canada STC reference C-10-1050 dated 18 December 2010/STC STC No. SA94-113 Issue 1 dated 28 October 2004/Aero Consulting Services Configuration Definition List D9542/20/Revision FX dated 14 December 2019/Instructions for Continued Airworthiness GC39660000-FC/Cen later revisions of the above...</td>
<td>Restricted to ASN 9369 and Refer to the Limitations sections of the required AFM Supplement and the referenced FCOM5.2.WLAN system is compatible only to IEEE 802.11(b).</td>
<td>21/12/2010</td>
<td>Active</td>
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<td>10033185</td>
<td>ROW 44, Inc.</td>
<td>REV. 3</td>
<td>FAA STC ST0253CH</td>
<td>B737-700/800-900-900E</td>
<td>Installation of Row 44 broadband antenna system provisions Revision 1: Transfer of ownership from Armstrong Aerospace to Row 44 Revision 2: Validates the configurations with Boeing's optional cabin routing life raft installed. EASA Certification basis formally upgraded from CS25 amdt. 1 to amdt. 2...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (a): H-01: Enhanced airworthiness programme for airplane systems - ICA on EWIS. The following paragraphs (a) at a later amendment: CS 25 amdt. 2 for...</td>
<td>- Master Data Index Listing (MDL) Doc. No. DC104-7023-00, Revision Y dated August 12, 2015/Instructions for Continued Airworthiness, Section 2, Airworthiness Limitations, as identified in Row 44 individual configuration Masterdata List contained within Master Data Index Listing, Document No. DC104-7023...</td>
<td>- The equipment for which these provisions are intended has been certified. Additional approval is required for the installation of this equipment and must be evaluated to assure satisfactory compliance with the applicable airworthiness standards/- For effective aircraft NOT fitted with Low Cabin A...</td>
<td>15/02/2017</td>
<td>Active</td>
<td>10033185</td>
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<tr>
<td>10033186</td>
<td>ROW 44, Inc.</td>
<td>REV. 6</td>
<td>FAA ST0256CH</td>
<td>-700/800-900-900ER</td>
<td>Installation and Activation of a Row 44 broadband antenna system provisions Revision 1: Transfer of ownership from Armstrong Aerospace to Row 44 Revision 3: Validation of the configurations with the option to install new ROW44 High Power Tran...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (a): H-01: Enhanced airworthiness programme for airplane systems - ICA on EWIS. The following paragraphs (a) at a later amendment: CS 25 amdt. 2 for...</td>
<td>- Master Data Index Listing (MDL) Doc. No. DC104-7024-00, Revision A dated December 28, 2016/Instructions for Continued Airworthiness as identified in each Master Data List referenced in the MDL Doc. No. DC104-7024-00, Revision A dated December 28, 2016...</td>
<td>- The Row 44 Wireless Broadband System is intended to provide internetconnection and email services to the airplane's cabin using portable electronic devices (PEDs). Any other use of this equipment requires additional approval. - Row 44 Broadband Antenna System Provisions installation must be accomplished...</td>
<td>07/11/2018</td>
<td>Active</td>
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<td>10033194</td>
<td>TELEDYNE UK LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B737-700/800-900-900ER</td>
<td>Installation of enhanced airborne data load/eIFV. 1: Transfer from Teledyne Limited to Teledyne UK Limited.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 17/13 Amdt 5 - EWIS Fire Protection, Wire Insulation flammability. CRH H-1 Enhanced Airworthiness Programme for Airplane Systems – ICA on EWIS. 2/0200/...</td>
<td>- Master Data List MOD-MDL-024-02 Issue 2 dated 12 Oct 2010/Instructions for Continued Airworthiness MOD-ICA-024 Rev. 1/10d12Feb 2010 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This modification is only applicable to 737 aircraft that already havem SAFD installed or have provisions for installation of an ADS-B compatibility must be established for each aircraft...</td>
<td>26/11/2020</td>
<td>Active</td>
<td>10033194</td>
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<tr>
<td>10033212</td>
<td>KUERZI AVIATION AG</td>
<td>REV. 1</td>
<td>MOD 500N</td>
<td>MDS20N Gyro camera &amp; equipment inst. (Gyrosys &quot;TURNKIT Assy&quot; for filming or live broadcasting)</td>
<td>CAR 6 dated December 20, 1996, including Amendments 6-1 through 6-6 andSpecial Conditions. The Certification Basis for the original product remains applicable to this certificated approval. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this...</td>
<td>- Flight Manual Supplement Doc. No. FMS 201032936; or later revisions of the above listed documents approved by EASA – Operating Instructions Doc. No. OPI-201032936; Master data list Doc. MDL-201032936;</td>
<td>Reg.: / S/N: LDN52; LN 068; LN0097/ Prior to installation of this design change it must be determined whether an interoperability or emissions levels of the original product with the new system.</td>
<td>14/02/2013</td>
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<td>AEROSPACE</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>AFM Supplement No. FMS10131.1, Rev. 1 dated 7 December 2010</td>
<td>None</td>
<td>04/01/2011</td>
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<td>TENENCIA Ltd.</td>
<td>Installation of Garmin G500/G600 PFD/ND</td>
<td>CS-23-1H-01</td>
<td>None</td>
<td>05/01/2011</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>Modification TEN 655, Avro 146-RJ85</td>
<td>Modification TEN 655, Avro 146-RJ85, Installation of 3R and 4R Galley and No 3 VHF COMM</td>
<td>Serial numbers E2323 and E2325 only.</td>
<td>05/01/2011</td>
<td>Active</td>
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<td>AEROSPACE DESIGN AND ENGINEERING CONSULTANTS LTD</td>
<td>Installation of First Class Seating</td>
<td>Modification Summary, MOD-123, Revision A dated 12/05/2010</td>
<td>None</td>
<td>05/01/2011</td>
<td>Active</td>
<td>10033237</td>
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<td>DART AEROSPACELTD</td>
<td>Automatic Door Opener Kits (Cabin and Baggage Doors).</td>
<td>MDL-D407-802 revision A dated 17 November 2010; Master Document List, ref. D407-802 revision A dated 13 October 2010</td>
<td>None</td>
<td>07/01/2011</td>
<td>Active</td>
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<td>Supreme Systems, Ltd.</td>
<td>The modification consists of removal of the Pratt &amp; Whitney PT6A-114/114A engine to install a Honeywell TPE331-12J engine and Hartnett propeller, identified below, along with associated installation components in accordance with Supreme Master Data List (SML) MOL-208-401. Rev. G, dated December 28, 202...</td>
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<td>208 ANG 208B</td>
<td>FAA Approval Manual Supplement, at the indicated revision level or later FAA-approved revision, are required for airplanes with the 900 SHP (-702TT or -704TT) engines installed. For 208 Model Landplane: Document SSL-SMS-208-12JR Landplane...</td>
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<td>208, 208B</td>
<td><strong>Limitations/Conditions</strong></td>
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<td>10033267</td>
<td>STC Compatibility with other STCs: Model 208 airplanes. Compatible with gross weight increase changeper STC SA700CH. Model 208 airplanes. Compatible with Waspair float modification STC SA3110GL. Waspair Service Kgf 25 must be installed on aircraft with serial number 20800329 and earlier per...</td>
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<td>F28 MK 0100</td>
<td>Prior to installation of this modification the airplane must be determined to be in accordance with the FAA policy for changes to the aircraft. Prior to installation of this modification it must be determined that the installation does not change the airworthiness of the product.</td>
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<td>SAFRAN ELECTRONICS &amp; DEFENSE/SURFACE</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s) CRI F-01 for Lithium Battery Instalation dated July 2003. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product...</td>
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<td>AS 355 NP</td>
<td>- Certification change N-93565 NP-$5/04-03. Installation Manual NP-$5/04-03. Maintenance Manual Supplement NP-$5/04-03. MDDL, Issue 10, dated May 2013. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product...</td>
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<td>SAFRAN ELECTRONICS &amp; DEFENSE/BUSINESS</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s) CRI F-01 for Lithium Battery Instalation dated July 2003. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product...</td>
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<td>8907-05223-2059, dated February 12, 2003 or later FAA-approved revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequences of this decision)</td>
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<td>208A2291AK</td>
<td>This modification is for the replacement of the Transponder Units and TCAS. This approval does not include the transponder and TCAS system, wiring, busses, antennas, and structural provisions, etc.</td>
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<td>208, 208, 208B</td>
<td>This modification is not compatible with: Cessna 208 series numbers 20800001 through 20800013, 20800416-20800500 and greater. Cessna 208B serial numbers 208B1110, 208B1116, 208B2000 and greater. FAA STC SA01478WIb. This modification has been verified to be compatible with both theoretical and...</td>
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<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>1003354</td>
<td>LEES AVIONICS Ltd.</td>
<td></td>
<td>LEARJET 35, LEARJET 35A</td>
<td>Installation of single or dual Universal Avionics UNS-1/1or UNS-1/1W FMS (Flight Management System)</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. Refer to Certification Plan CP-1159, issue 1, and Compliance Record CR-1159 for the requirements for environmental protection and the area...</td>
<td>Refer to Master Documents List MDL-1159, Issue 1 for later revisions of the above listed documents approved by EASA</td>
<td>17/12/2010</td>
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<td>1003397</td>
<td>NASCO AIRCRAFT BRAKE, INC.</td>
<td>REV.1</td>
<td>FAA STC ST0321LA</td>
<td>MD-88 DC-9/7 (MD-87) DC-9/8 (MD-83) DC-9/2 (MD-82) DC-9/1 (MD-81)</td>
<td>FAA STC No. ST0321LA Validation &quot;Installation and replacement of brake Components&quot;</td>
<td>Nasco MD-50/Compliance check list FAA project ST13341LA; Nasco Configuration Control Drawing Record of drawing changes Doc, NCC010 Revision NC, dated 10 September 2010, Nasco Component Maintenance Manual containing Instruction for Continued Airworthiness (ICA), Doc. CNM/MD0601, Revision IR, dated 30 Sep...</td>
<td>Refer to FAA STC ST0321LA</td>
<td>18/01/2011</td>
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<td>1003410</td>
<td>ROSEN SUNVISOR SYSTEMS LLC.</td>
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<td>FAA STC SA3529NM</td>
<td>DHC-6-100, -200, -300</td>
<td>Cockpit Sun Visor installation in accordance with FAA STC SA3529NM</td>
<td>Rosen Drawing List Number RTO-002L, Rev. A, dated June 25, 2010; Rosen Survivor Systems Installation Instruction, Doc. 9641-0199-001 Rev. A, dated June 29, 2010; 207014 later revisions of the above listed documents approved by EASA; EASA in accordance with FAA STC ST0321LA</td>
<td>none</td>
<td>19/01/2011</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 1障</td>
<td>FAA STC ST0947NY</td>
<td>ATR 72-212A, ATR 72-211, ATR 72-212, ATR 72-201, ATR 73-202, ATR 72-101, ATR 72-102, ATR 42-320, ATR 42-500, ATR 42-200, ATR 42-300</td>
<td>Installation of Delta Engineering Synchro's and Sagem EAD6 AFDAU with AFM (Aircraft Performance Monitor)</td>
<td>Delta Engineering Master Data List. ref: 0701-08413-2166 Revision 1 dated October 26th, 2013</td>
<td>1- The AFM (denied, included in Sagem ED6 AFDAU), is advisory only. Following the various configurations supported by this STC, whether the Delta Engineering Synchro's are interfaced with the Sagem ED6 AFDAU, whether the Delta Engineering Synchro's must interface with the Sagem ED 35 AFDAU component.</td>
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<tr>
<td>10033449 AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td>A319-133</td>
<td></td>
<td>Airbus Corporate Jet Centre Modification CJ0615 - &quot;VIP Interior - HNA&quot; Revision due to update of AFM-S and MPD-S</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative avionics requirements as listed in the General ORP G-413 &quot;ACJ C Project Certification Basis&quot;. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>-Modification Approval Sheet, ref: CJ0615 Issue 5, dated 24th October 2013 - Amendment Flight Manual: ref: FM-S MOD CJ0615 dated 1st October 2013 - Maintenance Planning Document - MPD Supplement, ref: MPD-S MOD CJ0615 dated 1st October 2013 or later revisions of the above listed document.</td>
<td>This STC is limited to A319-133MSN 4042 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a noise effect upon the airworthiness of the product. The instal...</td>
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<td>10033452 STC TWENTY ONE LIMITED</td>
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<td>B757-200</td>
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<td>Carriage of Horses &amp; Grooms in Freighter a/c</td>
<td>The Certification Basis for the original product and the following additional or alternative avionics &amp; environmental requirements are applicable to the aircraft. For changed areas, the Special Conditions are part of the defined Certification Basis.</td>
<td>Master Document List, S21.MDL-0964, Initial issue, 19/01/2011 - Airplane Flight Manual Supplement, S21.FMS-0071, Issue 2, 19/01/2011 - Instructions for Continued Airworthiness, S21.TEC-0288, Issue 1, 11/17/2012 or later revisions of the above listed documents approved by EASA.</td>
<td>21/01/2011 Active 10033452</td>
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<td>10033463 S-TEC CORPORATION</td>
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<td>FAA STC SAAH1SW-D</td>
<td>TB9, TB10, TB20, TB21, TB200</td>
<td>Installation of S-TEC System S6/5X Two Axis Automatic Flight Guidance System, Model ST-647 (28 Volt System)</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval.</td>
<td>1) Installation in accordance with both: FAA Approved S-TEC Bulletin No. 847, dated 02-Nov-09, Revision 4 - FAA Approved S-TEC Master Drawing List No. 92559, dated 02-Nov-09, Revision D2 - Operation in accordance with: a) For Socata models TB9, TB 10, and TB 200 (Group Aerospace) FAA Approved Pilot...</td>
<td>Optional Autopilot System not available on Socata model TB 9, TB 10, and TB 200 (Group Aerospace).</td>
<td>24/01/2011 Active 10033463</td>
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<td>10033486 HAE CO CABIN SOLUTIONS, LLC</td>
<td>REV. 1</td>
<td>FAA STC ST0134AT-D</td>
<td>747-400F 747-400, 747-400 (B9C)</td>
<td>Installation of a Rockwell Collins Aero H+ Satcom System in accordance with TIMCO Master Data List 04T098A,01, Rev. IR, dated November 25, 2005 or later FAA approved revision</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval.</td>
<td>N.A.</td>
<td>None</td>
<td>04/02/2011 Active 10033486</td>
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### List of EASA Supplemental Type Certificates

#### 10033488

**B/E AEROSPACE, INC.**

- STC Holder: B/E AEROSPACE
- Rev. 01/02/2012
- Model: 777-300, 777-300ER
- Description: Installation of B/E Aerospace Pantry Air Chiller P/N 470-1
- Certification basis: Certification basis
- Associated Technical Documentation: Master drawings
- Limitations/Conditions: Master drawings
- Date of Issue: 01/02/2011
- Status: Active
- System Description: TCCA STC

#### 10033490

**HELICOPTER AIR TRANSPORT GESELLSCHAFT & CO. KG**

- STC Holder: B/E AEROSPACE
- Rev. 03/07/2024
- Model: EC135 P1, EC135 P2, EC135 P3, EC135 P4
- Description: Replacement of Onboard Talon System with improved (OBS) stick AS 21-17 Rev. 1: Modification of double hook beam and release mechanism Rev. 2: Modification of the design of the Push-Pull Cablerelease System to optimize laverformance and flexibility Rev. 3: Adds wiring & Installations
- Certification basis: Certification basis
- Associated Technical Documentation: Master drawings
- Limitations/Conditions: Installation does not constitute operational approval Approval of PCDS (Personal Carrying Device System) for external load operation must be granted by the competent authority Prior to installation of this design change it must be determined that the interference between this design change and...
- Date of Issue: 01/02/2011
- Status: Active
- System Description: CAA UK

#### 10033501

**CARLISLE INTERCONNECT TECHNOLOGIES**

- STC Holder: B/E AEROSPACE
- Rev. 05/01/2021
- Description: Installation of ECS EZMount® mechanical mounting assembly (replacement of cockpit sliding window/window). 3 Addition of "New" models to the applicability list Rev. 4: Correction of models
- Certification basis: Certification basis
- Associated Technical Documentation: Master data list
- Limitations/Conditions: Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will introduce no harmful effect upon the airworthiness of the product The equipment for which these provisions are int...
- Date of Issue: 01/02/2011
- Status: Active

#### 10033551

**H4 AEROSPACE (UK) LTD**

- STC Holder: H4 Aerospace Modification
- Rev. 01/02/2012
- Model: 737-300, 737-400, 737-500, 737-800
- Description: Introduction of Cockpit Door Surveillance and Cabin Witness System
- Certification basis: Certification basis
- Associated Technical Documentation: Master data list
- Limitations/Conditions: None
- Date of Issue: 01/02/2012
- Status: Active

#### 10033559

**BOMBARDIER AEROSPACE**

- STC Holder: Bombardier Aerospace
- Rev. 10/08/2011
- Model: CL-600-2B16
- Description: Dual Power for Electronic Flight Bag (EFB) - Challenger 604/605
- Certification basis: Certification basis
- Associated Technical Documentation: Master data list
- Limitations/Conditions: The complete custom aircraft interior defined by EASA STC No. 100160/01 MUST be installed concurrently or as a pre...
- Date of Issue: 10/08/2011
- Status: Active
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<td>10033580</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>EC135 T3</td>
<td>Installation of Speedtech Nightvan XP on EC135 equipped with EC135 equipment post. Equipment Inspection and re-certification to close the requirement for AD No. 2010-0183R1. Note: Identiﬁc to STC with ref. 10035275, re-issued for administrative purposes.</td>
<td>The Certification Basis for the original product remains applicable tothis certiﬁcate/approval.</td>
<td>None</td>
<td>Maintenance Requirements Document MR/135787-001, Issue 0.</td>
<td>16/11/2010</td>
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<td>10033617</td>
<td>AIRFILM CAMERA SYSTEMS, INC</td>
<td>FAA STC SOR2297LA</td>
<td>3. AB412 AND AB412EP 1, BELL 212, 412 AND 412EP</td>
<td>Nose Mount Utility Bracket</td>
<td>The Certification Basis for the original product remains applicable tothis certiﬁcate/approval.</td>
<td>None</td>
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<td>10033622</td>
<td>KUERZI AVIATION AG</td>
<td>AS 350 B2, AS 360 B3</td>
<td>GS001 Avionic Retrofit (Option 1 &amp; 2)</td>
<td>The Certification Basis for the original product remains applicable tothis certiﬁcate/approval.</td>
<td>AF412NM-010 revision N/C dated 1 October 2009 - Master Drawing List;AF412NM-006 revision IF dated 25 May 2010 - Installation</td>
<td>None</td>
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<td>10033638</td>
<td>AERODATA AG</td>
<td>REV. 1</td>
<td>HAWKER 800 (U125) BAE.125 SERIES 800A</td>
<td>Installation of Aerodata FIS (Flight Inspection System).The FIS installation consists of non-essential sensor systems in the cabin and a camera system in the tail cone to enable calibrationsmissions for navigation equipment.</td>
<td>The Certification Basis for the original product remains applicable tothis certiﬁcate/approval.</td>
<td>None</td>
<td>Installation in accordance with Aerodata Master document list;MDL-IA C-155, Rev. 2 Operation in accordance with Aerodata Flight Manual Supplement, AFM-113 Rev. 1 Continued Airworthiness in accordance with Aerodata Aircraft Maintenance Manual Supplement, AMM-1045, Rev 0 for later revisions of the above list...</td>
<td>25/02/2011</td>
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<td>10033641</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>EMB-135BJ</td>
<td>Modification 289A. Thrane &amp; Thrane Aero SB Lite Satcom Installation</td>
<td>The Certification Basis for the original product remains applicable tothis certiﬁcate/approval.</td>
<td>None</td>
<td>Modification 289A &quot;Thrane &amp; Thrane Aero SB Lite Satcom Installation/ Master Document List 289A-MDL, Rev. 0, dated 20-07-11, AFM Supplement 289A-AMM Rev. N/C, dated 16-10-09, AMM Supplement with Airworthiness Limitations 289A-AMM Rev. C, dated 24-01-11 for later revisions of the above listed documents approved...</td>
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<td>Limitations/Conditions</td>
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<td>AELLYPER S.A.</td>
<td>EC135 P2, EC135 P2+</td>
<td>Installation of STAR SAFIRE HD Camera System &amp; Digital Transmission System</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this STC/approval JAR 27 Amendment 3, dated 01/04/2002, for paragraphs 27.561 &amp; 27.602 JAR 27 Amendment 2, dated 01/05/2001, for paragraphs 27.610, 27.605 &amp;…</td>
<td>None</td>
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<td>AVIATION SUPPORT GmbH</td>
<td>MD900</td>
<td>Installation of an Equipment Rack</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this STC/approval CS-27 Amendment 2. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product must be determined.</td>
<td>Master drawing list: MDL-00-0906 Rev 0-0-0; Compliance Report CR-ASE-00096 Rev 0-1; ICA-ASE-00096 Rev 0-2; TRM Supplement; FMS-ASE-00096 Rev 0-2 Rev 0-2 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>02/02/2011</td>
<td>Active</td>
<td>10033667</td>
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<td>AVIATION SUPPORT GmbH</td>
<td>AB206 A, AB206 B</td>
<td>Installation of ALMA G2 system (Airborne Laser Methane Assessment) – Gas Detection System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 27.25, CS 27.27, CS 27.29, CS 27.31, CS 27.601, CS 27.603, CS 27.605 (+), CS 27.607, CS 27.609, CS 27.610, CS 27.611, CS 27.613, CS 27.615, CS 27.711 (+), CS 27.713, C…</td>
<td>FMS-ASE-00092 revision 0-2 dated 17 August 2015</td>
<td>18/06/2015</td>
<td>Active</td>
<td>10033670</td>
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<td>LIHTHANSA TECHNIK AG</td>
<td>DHC-8-401, -402</td>
<td>Installation of non-electrical Floor Path Marking System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Change 9 and FAR 25.1303(d), amended 12/31 for environmental protection and the associated certified noise and/or emissions level of the original product.</td>
<td>Installation must be in accordance with LHT Supplemental Type Definition Document STDD33-1001M Issue 01</td>
<td>Not Applicable</td>
<td>17/01/2014</td>
<td>Active</td>
<td>10033691</td>
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<td>SAAB AB</td>
<td>AS 355 N, AS 355 NP</td>
<td>Installation of new TopEye Mission Box</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>25/10/2023</td>
<td>Active</td>
<td>10033697</td>
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List of EASA Supplemental Type Certificates

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<th>STC No.</th>
<th>STC Holder</th>
<th>Description</th>
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<th>System Record</th>
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<tbody>
<tr>
<td>10033707</td>
<td>AERONAUTICAL ACCESSORIES, INC.</td>
<td>FAA STC SH121810 206A-3, 206L-4, 407 206A, 206B, 206L, 206L-1</td>
<td>Maintenance Step</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>04/02/2011</td>
<td>Active</td>
<td>10033707</td>
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<tr>
<td>10033710</td>
<td>LUFTHANSA CERTIFICATION SERVICES</td>
<td>TCCA STC NO. SA4-113 ISSUE 1 BD-700-1A10 Installation, Complete Custom Aircraft Interior - ASN 9385</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada SuC reference C-10-2758 dated 02 February 2011; TCCA STC No. SA4-113 Issue 1 dated 28 October 2009; Aeroc Consulting Services Configuration Definition List DD54000 Revision/GB dated 28 January 2011; Instructions for Continued Airworthiness GC93850000-FICA or later revisions of the above.</td>
<td>Restricted to ASN 9385 and.</td>
<td>04/02/2011</td>
<td>Active</td>
<td>10033710</td>
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<td>10033724</td>
<td>BRISTOL AEROSPACE LTD.</td>
<td>STC SH10-4 429 Installation of Wire Strike Protection System Revision 1: Administrative Revision to and details in associated Technical Documentation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EM106900 issue D dated 18 October 2010 - Modification Data Summary; PM-965-067 issue G dated 14 October 2010 - Instructions for Continued Airworthiness; or later revisions of the above listed documents(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU...</td>
<td>Prior to installation of this modification it must be determined that the relationship between this modification and any other previously installed modification and/or repair will introduce no adverse effects upon the airworthiness of the product.</td>
<td>09/05/2019</td>
<td>Active</td>
<td>10033724</td>
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<tr>
<td>10033738</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>STEC 8747-400 Activation of ADS-B Out capability via 1090MHz Extended Supplier Interconnect System inaccordance with Lufthansa Supplemental Type Definition Document STD34-103M Issue 01, Rev 02, dated Feb. 10, 2011 or later approved/revision.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Doc. No. A-05-10225 Rev IR Airplane Flight Manual Supplement/Doc. No. A-05-10225-ICA Issue 1, Rev. IR Instructions for Continued Airworthiness; or later revisions of the above listed documents approved by EASA</td>
<td>none</td>
<td>17/02/2011</td>
<td>Active</td>
<td>10033738</td>
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<td>10033754</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC ST00790CH Installation of EZMount(TM) Arm mechanical mounting assemblies on lowerframe of cockpil sliding window in accordance with Electronic Cable Specialists Master Data List ECS-207476 Rev. A, dated March 3, 2009; earlier FAA approved revisions. Validation of FAA STC ST00790CH</td>
<td>Pre FAA TCDS A22WE for parts not changed or affected by this change. The Certification Basis for the parts changed or affected by this change is the reference data of application February 4, 2008 is shown on TCDS A22WE. In addition, the applicant has complied with 14 CFR Part 21.26 amended by amendment...</td>
<td>AFM Supplement as listed in Appendix A of ECS Master Data List ECS-207476 Rev. A</td>
<td>AFM Supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-207476. Rev. A, dated March 4, 2009 or later FAA approval revision, is required on board the modified aircraft. The equipment for which these provisions are intended is not part of the aircraft. Additional EASA...</td>
<td>12/08/2013</td>
<td>Active</td>
<td>10033754</td>
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<tr>
<td>10033764</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>REV. 1 P-68TC &quot;OBSERVER&quot; Change of cabin layout. For the installation of different mission equipment in a VuelanP-68TC aircraft, the aeroshell has to be modified and provisions In-accordance Mission Equipment have to be installed. This modification includes the removal of three cabin seats. Per secessions on aircrafts...</td>
<td>Elected to comply: CS 23.23, Amdt. 2 and FAR 23, effective 1 Apr 2010. 180698c 23.337, 23.341 and 23.581 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions level...</td>
<td>Engineering Order ABP067-E4-2590-01-03, Issue FAA Supplement ABP067-AM-2590-01-02, Issue Aeronautical Consultative Definition ABP068-DE-2590-02-02, Issue 2 Technical Order ABP067-TO-2460-01-01, Issue 21.26 later revisions of the above listed documents approved by EASA</td>
<td>Limited to S/N 454, 456 and 459</td>
<td>04/01/2012</td>
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<td>STC Holder</td>
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<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>TENENCIA Ltd.</td>
<td>CA-A Uk</td>
<td>F27 MK 050 &amp; F27 MK 0502</td>
<td>Installation of a Single UHNS-1W Universal Avionics Flight Management System.</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Plan TEN/FON/0004 Issue 1 Statement of Compliance and Report of Statement of Compliance D/ENG/0014 Issue 7 System Safety Assessment 387-01-0008 Revision B/ Master Drawing List 298-00-001-01 of 31 March 2009/AFA</td>
<td>Report</td>
<td>03/02/2011</td>
<td>Active</td>
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<tr>
<td>ACS-NAI Ltd.</td>
<td>TCCA STC NO.3-C-LSA11-040/D</td>
<td>B700-1A11</td>
<td>Cabin Modifications for Wheelchair Access.</td>
<td>EASA M.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA SoC ref. C-10-1061 dated 08 February 2011/ TCCA STC NO. C-LSA11-040/D Issue 1 dated 02 February 2011/ACS-NAI Ltd. Modification Data Summary Q59000 Revision NC or later approved revisions or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF.</td>
<td>Restricted to ASN 9573 Previous embodiment of EASA STC No. 10031088 (TCCA STC SA-05-60).</td>
<td>09/02/2011</td>
<td>Active</td>
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<td>B/E AEROSPACE, Inc.</td>
<td>FAA STC STO3282AT</td>
<td>737-600/-700/-800/-900</td>
<td>Installation of Dual Collins HFS-900 High Frequency (HF) communications system. Revision 1. Transfer of EASA STC 10033802 from EMTEQ ENGINEERING to B/E AEROSPACE.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (a)(1) “Enhanced Airworthiness Program for Airplane Systems – ICA on EWIS” Issue 2. The requirements for environmental protection and the associated certificate...</td>
<td>EMTEQ Master Data List (MDL) Document No. 2-36870-01W Revision 1 dated January 10, 2011/EMTEQ Airplanes Flight Manual Supplement (AFMS) Document No. 2-36870-049 dated June 18, 2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF or subsequent.</td>
<td>Prior to installation of this design change it must be determined that the teamwork/integration between this design change and any other previously installed design changes or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/08/2015</td>
<td>Active</td>
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<td>FR AVIATION Ltd/DBA COBHAM AVIATION SERVICES UK</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B300</td>
<td>Observer Training Platform Mod.-B300. CS-23.B Installation must be in accordance with Master Document List included in the Design Change Statement DC-0166 Iss.1 dated 05 Jan 2011 or later approved or latest revision. Operation must be in accordance with Pilots Operating Handbook Supplement No. FRAHPAM0120 Rev. 2. Dated 18th January 2011 or subsequent.</td>
<td>The approval is limited to the Aircraft Serial Numbers: FL-597&lt;-&gt;FL-597&lt;-&gt;FL-597&lt;-&gt;FL-597.</td>
<td>The approval is limited to the Aircraft Serial Numbers: FL-597&lt;-&gt;FL-597&lt;-&gt;FL-597&lt;-&gt;FL-597.</td>
<td>The approval is limited to the Aircraft Serial Numbers: FL-597&lt;-&gt;FL-597&lt;-&gt;FL-597&lt;-&gt;FL-597.</td>
<td>09/02/2011</td>
<td>Active</td>
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<td>AIR ALLIANCE GmbH</td>
<td>PA-42-1000</td>
<td></td>
<td>Installation of SAT antenna (Data-Link) into pressurized fuselage.</td>
<td>The Certification Basis of the original product and the following additional or alternative airworthiness requirements are applicable to this change. For the area affected by the change CS-23 has been selected to comply with where possible. The certificate noise and/or emissions levels of the original product...</td>
<td>EASA approved Master Document List. Doc. No. 2010_040_PM42_AF02 Form 2 Master Doc List. Revision 1, approved on 10-Jan-2011/Engineering Order. Doc. No. 2010_040_PM42_500. Revision 1, released 2010/Instructions for continued Airworthiness, Doc. No. 2010_040_PM42_500, Rev. 0, released 27-Sep-2010 or later...</td>
<td>None.</td>
<td>10/02/2011</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
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<td>Description</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10033819</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>777F</td>
<td>Installation of Rosemount Aerospace EFB Systems.</td>
<td>The Certification Basis for the original product as amended setting forth one additional or alternative airworthiness requirement. The following paragraphs at a later amendment CS-25.305A (4) is in case of applicable MDR, configuration A and B for 25.305A(c)(5) 25.3(3) and 25.305A(c)(5) are 25.337(c) 25.337(e) and 25.337(f).</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair document will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/05/2017</td>
<td>Active</td>
<td>10033819</td>
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<td>10033820</td>
<td>S-TEC CORPORATION</td>
<td>PA-31, PA-31-300, PA-31-325</td>
<td>Installation of S-TEC 20/30 Single and Two Axis Automatic Flight Guidance Systems, Model ST-727-2030 (28V - System)</td>
<td>The Certification Basis of the original product remains applicable without change. The certification noise levels or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>FAA approved S-TEC Bulletin No. 627, Revision 4, dated 26-Apr-2001, with approved revision FAA approved S-TEC Master Drawing List No. 921529, Revision D, dated 08-Apr-2001, later approved revision FAA/DAS approved Pilot's Operating Handbook and/or Aircraft Flight Manual Supplement, P/N 951538.</td>
<td>10/02/2011</td>
<td>Active</td>
<td>10033820</td>
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<tr>
<td>10033835</td>
<td>DASSAULT FALCON SERVICE</td>
<td>FALCON 900EX</td>
<td>Falcon 900EX aircraft serial numbers 065 and 073 - Completion Definition. Revision 1: to remove limitation on S/N 073 further to - compliance of Passenger seat cushions (Qty 8 seats P/N N2532-14A000 with JAR 25.863 (b) (s), approved under minor change FDFS9557 Issue B, and - LH Divan cushion...</td>
<td>The Certification Basis for the original product remains applicable without change. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC FDFS9557-CA5 Issue B dated 02 June 2017.</td>
<td>25/01/2018</td>
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<td>EASA STC No.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10033844</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>REV. 2</td>
<td>T182T</td>
<td>Camera Hatch Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval FAA Certification Basis 3A3: FAR 23 effective February 1, 1995, as amended y 23.1 through 23.6 except as otherwise defined in the TCDS. However the applicant elected to comply with CS-23 amendment 7. The r...</td>
<td>none</td>
<td>none</td>
<td>24/05/2012</td>
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<tr>
<td>10033847</td>
<td>JCB AERO SA</td>
<td></td>
<td>A320</td>
<td>Installation of Hi-Low Table for VIP cabin interior</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged.</td>
<td>none</td>
<td>none</td>
<td>14/02/2011</td>
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<td>10033865</td>
<td>SKYTRAC SYSTEMS LTD.</td>
<td></td>
<td>TCCA STC NO. SH10-24</td>
<td>EC 135 T2(CPDGS),EC 135 T2+, EC 135 T1(CPDGS),EC 135 T1 PD2+, EC 135 P1(CDGS),EC 135 P1CPDS</td>
<td>Sky Track ISAT AIRBORNE POSITION / COMMUNICATION UPGRADEThemodification upgrades an existing ISAT-100 Aireon platform with new ISAT-200 software</td>
<td>Aeronica Design Systems LTD. Master Drawing List, ref. doc. A320ALL01-MDL-01, Revision B, dated 14th February 2011 - Project Data List, ref. doc. A320ALL01-PDL-01, Revision A, dated 7th February 2011 - Declaration of Compliance for Project A320ALL01, dated 11th February 2011 for later revisions of the above listed documents approved by EASA</td>
<td>none</td>
<td>15/02/2011</td>
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<td>10033870</td>
<td>AIRBUS HELICOPTERS DEUTSCHLAND GmbH</td>
<td>(EASA.R.S.01568)</td>
<td>EC 120 B</td>
<td>Installation on an EC120B helicopter of a ETSO-certified ETOS C113(Moving Map system (MT VisionAir EP))</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged.</td>
<td>none</td>
<td>None</td>
<td>15/02/2011</td>
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<td>10033883</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
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<td>407</td>
<td>Installation of Antenna Mount on Tailboom</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged.</td>
<td>none</td>
<td>none</td>
<td>16/02/2011</td>
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<td>10033889</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA STC NO. SA11-1 ISSUE 1</td>
<td>CL-600-2B16</td>
<td>Activation of Wireless Local Area Network Access Point ICD NXT Maritime System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval FAA Certification Basis 3A3: FAR 23 effective February 1, 1995, as amended y 23.1 through 23.6 except as otherwise defined in the TCDS. However the applicant elected to comply with CS-23 amendment 7. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged.</td>
<td>none</td>
<td>none</td>
<td>16/02/2011</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10033895</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC ST-REV004RC-D</td>
<td>EC 135 T2 (CPS2) EC 135 T2+ EC 135 T1 (CPS2) EC 135 P2 (CPS2) EC 135 P3+ EC 135 P1 (CPS2) EC 135 P1 (CPS2)</td>
<td>Installation of a collective activated tenth hourmeter in accordance with AEC Drawing List 135A02-8008. This EASA STC is based on FAA STC SR904RC-D.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>- AEC Drawing List 135A02-8008, Revision D, dated 03/01/2016 Instructions for Continued Amortization 135B85-006, dated 05 June 2007 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>17/02/2011</td>
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<td>10033896</td>
<td>AIRCRAFT DESIGN CERTIFICATION GmbH</td>
<td>REV. 2</td>
<td>VENTUS CM</td>
<td>Steerable Tail Wheel</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Annex to the Flight Manual Ventus dl, issue 25 January 2011 Panel Aeronautics Maintenance manuals, Annex for Scheppe-h-Hirth Vertusdh, equipped with ADAC- DC-38-001 steerable tail wheel; Anhang zum Flughandbuch Ventus dl, Ausgabe 25 Januar 2011 Motorenpf. Wartungshandbuch, Anhang Scheppe-Hirth Ve...</td>
<td>None.</td>
<td>Ventus dl with Schemp-Hirth technical Note 825-6.</td>
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<tr>
<td>10033897</td>
<td>U.S. TECHNICAL CONSULTANTS, INC.</td>
<td>FAA STC STO335LA</td>
<td>B767-200, -300, -300F, -400ER</td>
<td>Installation of Class 2 Electronic Flight Bag (EFB) provisions.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Any equipment that is connected to the electrical power provisions/certified by this STC must be tested in accordance with navAero's Electrical and Environmental Substantiation for Lithium-based battery in navAero icdg C22 EFB System Document No. E-1525-11-0325-00, revision C, 0...</td>
<td>None.</td>
<td>None.</td>
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<tr>
<td>10033913</td>
<td>TATENHILL AVIATION Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Design Data for the Approval of a Honeywell KFD840 Primary Flight Display (PFD).</td>
<td>CS-23 (The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Manual M. 00910 Revision DMaster Document ListMDL.00910 Revision Altimetry Instructions for Continued Airworthiness ICA. 00910 Revision AirFlight Manual Supplement FMS. 00910 Revision Air later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>None.</td>
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<td>10033914</td>
<td>TELEDYNE CONTROLS, LLC</td>
<td>FAA STC STO1718A</td>
<td>A310-304, -312, -324, -326 A310-203, -204, -221, -222 A300F4-605R, -620R A300B4-601, -603, -605R A300B4-2C, -103, -203 A300B4-3C, A300B2-1A, -1C, -203</td>
<td>Installation of Wireless Ground Link - Quick Access Recorder (WQAR) - This STC consists in the installation of the WQAR on Airbus A300/A310/A300-600 series.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, Ref. MDM.2243800-31X, dated 14 March 2005/Instruct for Continuing Amortization, Ref. ICA.2243800-A1-1, dated 29 February 2005, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>This STC is applicable to the following WQAR p/n (2243800-xxx): 2243800-21- 2243800-31, 2243800-33- 2243800-35, 2243800-37- 2243800-39- 2243800-42, 2243800-43, 2243800-44, 2243800-62, 2243800-63, 2243800-64, 2243800-71, 2243800-73, 2243800-75, 2243800-77, 2243800-79- 2243800-81; 2243800-91.Prior to...</td>
<td>12/04/2016</td>
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List of EASA Supplemental Type Certificates
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<th>STC Holder</th>
<th>STC No.</th>
<th>STC holder Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<tr>
<td>AVIONIK STRAUSSING ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td>10033957</td>
<td>3202,3306,3307,3212,3217,3109,3110,3112,3116,3201</td>
<td>Dual/Single Installation of Garmin GNS430/AJW I GNS530/AJW and optional installation of Marker Receiver Honeywell (Bendix/King) KMR675</td>
<td>The Certification Basis for the original product and the following:</td>
<td>Instructions for Installation: A&amp;R-2010-06 1-F1.10.00, Rev. 0</td>
<td>Flight Manual Supplement: A&amp;R-2010-06 1-F1.10.00, Rev. 0</td>
<td>none</td>
<td>24/02/2011</td>
<td>Active</td>
<td>10033957</td>
</tr>
<tr>
<td>ROEDER Pnakocion GmbH</td>
<td>REV. 1</td>
<td>10033946</td>
<td>421A/421B/421C/425 402C/411/411A/411/44/41/41/41/4/ 421/401/401A/401B/402/402A/402B</td>
<td>Installation of a Traffic Advisory System Avidyne TMS10420.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/</td>
<td>Engineering Order C421- NY-08167-05, Rev. 1 Maintenance Procedures C421-NY-08167-07, Rev. 1 Flight Supplement C421-NY-08167-07, Rev. 1 for later revisions of the above listed documents approved by EASA.</td>
<td>A compatible Multi Function Display (MFD) must have been already installed. Prior to installation of this modification it must be determined that...</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10033946</td>
<td></td>
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<tr>
<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td>REV. 1</td>
<td>10033951</td>
<td>STC 3005/10 revision 0 - Modification of installation des équipements radio eBRAINAV avec couplage au pilote automatique.</td>
<td>FR182</td>
<td>The Certification Basis for the original product remains applicable to this certificate/</td>
<td>Supplement Manual de Vol SEAM/SMV-3005/10 Revision 0 dated 22 April 2011 for later revisions of the above listed documents approved by EASA.</td>
<td>Limitation to Cessna FR182 equipped with ASPEN EFD 1000 and NAVOMATIC300A.</td>
<td>22/02/2011</td>
<td>Active</td>
<td>10033951</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>10033960</td>
<td>B200/B300</td>
<td>Provisions for and installation of a SLAR and a Homer antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/</td>
<td>Engineering Order, Doc. No. 2003299-EO6(&lt;r&gt; 2, dated 19 April 2018(&lt;r&gt; 2, &lt;r&gt; or later approved issue Aircraft Flight Manual Supplement, Doc. No. 2003299-AM8, Issue 1, dated 14 November 2017, or later approved issue Issue for Continued Airworthiness, Doc No. 2003299-CA8, Issue 1, dated 15 Nov...</td>
<td>The installation of this modification by third persons is subject to the permission of the approval holder and disposal of the approvable documentation. Prior to installation of this modification it must be determined that the integration of this modification and any other prev...</td>
<td>02/06/2018</td>
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<td>STC Holder</td>
<td>Model</td>
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<td>ROEDER PRAEVISION GMBH</td>
<td>421A/421B/421C/425 402C/411/411A/414/414A/427 401/401A/401B/402/402/402A/402B</td>
<td>Single installation of Garmin GNS 430 (A) W/GNS 530 (A/W and optional installation of Roll Steering Converter GDC31.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Engineering Order C21-NY-08168-05, Rev. 2 Flight Manual Supplement: C21-TY-08168-08a, Rev. 2; C21-NY-08168-08b, Rev. 1 Maintenance Procedures: C21-NY-08168-07, Rev. 1 for later revisions of the above listed documents approved by EASA.</td>
<td>An independent CMM or MVI DME or AD/FDM must have been already installed. Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10033970</td>
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<td>CRANE AEROSPACE &amp; ELECTRONICS</td>
<td>LEARJET 60</td>
<td>Installation of Tire Pressure Sensor FAA STC ST0316LA (Date of issuance: September 28, 2010)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Refer to Master Data List 83-216-01, rev. N for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions/this decision).</td>
<td>None</td>
<td>23/02/2011</td>
<td>Active</td>
<td>10033973</td>
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<tr>
<td>SAFRAN ELECTRONICS &amp; DEFENSEBUSINE</td>
<td>A109E, A109S, AW109SP A109E, A109S</td>
<td>Flight Data Monitoring and Transmitting System HELICOM V1, HELICOM V2M and HELICOM V2+ NUM Revision 1: Change in system name (Safetyplane became HELICOM V1), new v2 version and associated documentation.Revision 2: Introduction of new components, new locations for antennae extends eligibility sof...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- RFMS N11-A109E_S_HV2 HV2+NUM version E, edition A, rev 0 dated 10th June 2016 - Master Document List M05-A109E_S HV2 HV2+NUM version E, edition A, rev 0 dated 14th June 2016 - Installation Manuals N7-A109E HV1 version E, edition A, rev 0 dated 14th June 2016 - Installation Manuals N7-A109E HV1 version E, edition B, rev 02-08 dated 14th June 2016... Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/10/2020</td>
<td>Active</td>
<td>10033974</td>
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<td>POLICE AVIATION SERVICES Ltd.</td>
<td>EC 635 T2+ EC 635 T1 (CPDS) EC 635 P2+ EC 635 T2 (CPDS) EC 135 T2+ EC 135 T1 (CPDS) EC 135 T1 (CD5) EC 135 P2+ EC 135 P1 (CD5) EC 135 P1 (CD6)</td>
<td>Incubator Installation EC 135T2 of EASA STC No 2004 3846 (dated 16/04/2004) from PAS Ltd to Bond Engineering Order C21-28/02/2004; dated 16 April 2004/Rev. PAS modification PAS/03/0488.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>CAA 2.A12NE 0089, EASA Approval number 2004-03849, dated 16 April 2004/Rev. PAS modification PAS/03/0488.</td>
<td>28/02/2010</td>
<td>Active</td>
<td>10033979</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>GV-SP</td>
<td>Aircraft Serial No 5302 completion and cabin interior installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificate/noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC522046210, Revision A, dated January 24, 2011, (or later approved revisions). Operation to be in accordance with Gulfstream Flight Manual Supplement GC52204M091 - Original Release, dated January 25, 2011, or later.</td>
<td>GV-S/P Serial No. 5302 Only</td>
<td>24/02/2011</td>
<td>Active</td>
<td>10033988</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>GV-SP</td>
<td>Aircraft Serial No 5264 completion and cabin interior installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificate/noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC522046203, Revision A, dated December 7, 2010, (or later approved revisions). Operation to be in accordance with Gulfstream Flight Manual Supplement GC52204M095 - Original Release, dated September 5, 2010, (or later...</td>
<td>GV-S/P Serial No. 5264 Only</td>
<td>04/03/2011</td>
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<td>EASA STC No</td>
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<td>Associated Technical Documentation</td>
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<td>10033995</td>
<td>BOMBARDIER INC</td>
<td>TCCA STC</td>
<td>SAS04-113, ISSUE 1</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9380.</td>
<td>EASA 8.A.009(The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.)</td>
<td>Transport Canada O/S reference C-10-0482 dated 17 February 2003.</td>
<td>Restricted to ASN 9390 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced ECOM10, 2. WLAN system is compatible only to IEEE 802.11(b).</td>
<td>24/02/2011</td>
<td>Active</td>
<td>10033995</td>
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<tr>
<td>10033998</td>
<td>BOMBARDIER INC</td>
<td>REV. 1</td>
<td>TCCA STC</td>
<td>SAS05-60</td>
<td>Alt Lavatory Backsplash replacement Installation, Complete Custom Aircraft Interior - ASN 9405</td>
<td>EASA TCDS No 8.A.009(The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.)</td>
<td>TCCA STC No 8.A.045-60 Issue 1 dated 14 June 2005.</td>
<td>Restricted to ASN 9405.</td>
<td>29/06/2011</td>
<td>Active</td>
<td>10033998</td>
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<tr>
<td>10034024</td>
<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td>EASA APPROVED</td>
<td>MODEL LIST</td>
<td>Dual Installation of Garmin GNS 430 (A) W/INS 530 (A)(TAWIS)</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval. EASA CS-23, Amendment 2 (selected to comply by the applicant, not required by 21.A.101) The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Engineering Work Order: ...EW-1126-340000 ; Rev. Flight Manual Supplement: ..., PO-1126-340000, Rev. Maintenance Manual Supplement: ..., MM-1126-340000, Rev. for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>16/02/2021</td>
<td>Active</td>
<td>10034024</td>
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<tr>
<td>10034031</td>
<td>APICAL INDUSTRIES, Inc.</td>
<td>FAA</td>
<td>SR00411LA</td>
<td>600N</td>
<td>Emergency Floats</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List MD600-1, Revision B, dated September 30, 1997 plus: Report A600N-3, Rev. A, dated 03/31/09, Report A600N-1, Rev. C, dated 03/31/09, Rotorcraft Flight Manual Supplement FMS-600(1) Rev C - Instructions for Continued Airworthiness ICA600-1, Rev. dated 11/03/09 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>01/03/2011</td>
<td>Active</td>
<td>10034031</td>
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<td>10034052</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 2</td>
<td>757-200</td>
<td>Introduction of IFE System Circuit Breaker, Relays and Wiring, 2. Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval. Changes: June 11, 12 (new STC holder). Requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AES Master Data List AES-757-1072MDL, Issue 3, 18/09/2011 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>27/04/2021</td>
<td>Active</td>
<td>10034052</td>
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<tr>
<td>10034074</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>525A</td>
<td>525</td>
<td>Installation of High Speed SATCOM system with on-board WLANInterface.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval. CRI F-101 (MOC) “WLAN Implementation in Cessna 525, 525A” The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Engineering Work Order: RUAG Doc.No. 355A-E0, Rev. -Flight Manual Supplement: RUAG Doc.No. 355A-A0M, Rev. -Maintenance Manual Supplement: RUAG Doc.No. 355A-AMMS Rev. -or later revisions of the above listed documents approved by EASA.</td>
<td>STC limited to S/N 525-0516 The WLAN clients (Handsets, Laptops, etc) are not covered by this STC. The usage of these components is requiring operational approval.</td>
<td>04/03/2011</td>
<td>Active</td>
<td>10034074</td>
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<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
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<td>EUROCOPTER UK Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>EC 130 B4</td>
<td>Installation of Sagem Cockpit Display System MH1/130/240 and revisions to Eurocopter UK EC130B4 Supplement No. 1</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and are applicable to this certificate/approval.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Restricted to ASN 9304 and I. To Limitations of the sections of the required AFM Supplement and the referenced FCOMS 2, WLAN system is compatible only to IEEE 802.11(b).</td>
<td>04/03/2011</td>
<td>Active</td>
<td>10034086</td>
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<tr>
<td>C.A. FLOR FLORDESIS</td>
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<td>FAA STC STO796LA</td>
<td>B737-300, -400, -500 SERIES B737-100, -200, -200C SERIES</td>
<td>Upgrade Flight Data Recorder System Parameters</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and are applicable to this certificate/approval.</td>
<td>Master Drawing List document No. FF-050209FDL, Rev. B dated 2 April 2005 for revision of the above listed documents/approved by EASA in accordance with EASA ED Decision 2004/004/CF (or subsequent revisions after this decision)</td>
<td>Not Applicable</td>
<td>08/03/2011</td>
<td>Active</td>
<td>10034098</td>
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<td>APICAL INDUSTRIES, Inc.</td>
<td>REV. 1</td>
<td>FAA STC SR02291LA</td>
<td>BELL 407</td>
<td>Cable Cutters Kit</td>
<td>The Bell 407 Cable Cutter kit provides a means of protecting the helicopter during a wire strike by providing the capability to cut cable or wire before catastrophic damage occurs.</td>
<td>Cable Cutters Kit The Bell 407 Cable Cutter kit provides a means of protecting the helicopter during a wire strike by providing the capability to cut cable or wire before catastrophic damage occurs.</td>
<td>None.</td>
<td>21/03/2011</td>
<td>Active</td>
<td>10034100</td>
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<tr>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 1</td>
<td>CL-600-2B16/ CL-600-2B19</td>
<td>Installation of a combined RDIUM-GPS Antenna as provision of operational Sky Tracking System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and are applicable to this certificate/approval.</td>
<td>1. ACC Master Document List ACC-CRL200-009I6 Revision 0 dated 24 February 2011, 2. ACC ICA CRL200-009I6-810-MM Revision 0 dated 29 November 2011 for installation of the above listed documents approved by EASA.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the aerodynamic of the product.</td>
<td>22/10/2018</td>
<td>Active</td>
<td>10034112</td>
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<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td>PIPER PA-31-350 (CHIEFTAIN)</td>
<td>Provision for Camera Installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and are applicable to this certificate/approval.</td>
<td>Installation in accordance with Master Record List Document No. 71335MR 5300 10 01B Operation in accordance with Supplement to Pilot Operating Handbook/POH Document No. 71335-AM 5100 10 01B Maintenance in accordance with Instructions for Continued Airworthiness No. 71335 CA 5300 10 01Bor later revs.</td>
<td>This STC revision includes only the structural changes. It does not mandate any installation of equipment (e.g. Laser scanner, computer, cameras). Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the aerodynamic of the product.</td>
<td>10/08/2018</td>
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<td>AIM Aviation (Henshalls) Ltd</td>
<td>AIM Aviation (Henshalls) STC Definition Document, DD 1189, Issue New, 04/02/2011 Instructions for Continued Airworthiness, ICA/PA7802/Rev, New, Feb 2011 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>None</td>
<td>14/03/2011</td>
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<td>10034165</td>
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<td>AIM Aviation (Henshalls) Ltd</td>
<td>AIM Aviation (Henshalls) STC Definition Document, DD 1189, Issue New, 04/02/2011 Instructions for Continued Airworthiness, ICA/PA7802/Rev, New, Feb 2011 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>None</td>
<td>14/03/2011</td>
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<td>CAA UK RESPONSIBIL ITY</td>
<td>FAA STC ST02863CH KAMAN K-1200</td>
<td>Approval of Rotorcraft Flight Manual Supplement FAA approved Rotorcraft Flight Manual Supplement for the Kaman Model K-1200 Helicopter, KATI-FM-632, Revision A, FAA-approved on 17th May 2007 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>21/03/2011</td>
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<td>10034129</td>
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<td>EASA in accordance with EASA ED Decision 2004/04/CFPL</td>
<td>FAA STC SR01561SE</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>None</td>
<td>21/02/2011</td>
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<td>B767-200</td>
<td>Installation of a Crew Galley in the G1A Position on a FreighterConversion a/c</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product unaltered and remain applicable to this approval.</td>
<td>None</td>
<td>14/03/2011</td>
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<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>A330-300</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>SR Technology Master Document List, ref: S.ISMD.100001-01 MDL dated 14 March 2011</td>
<td></td>
<td>1. This STC must be implemented together with EASA STC ref: 10031808Rev 01 (and associated Approved Model List), or later revisions approved by EASA.</td>
<td>20/04/2021</td>
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<td>10034187</td>
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<td>BOMBDIER INC.</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9391</td>
<td>EASA M.909 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
<td>Restricted to ASN 9391 and 1. Refer to the Limitations sections of the required APM Supplement and the referenced FCOM. The WLAN system is compatible only to IEEE 802.11(b).</td>
<td>15/03/2011</td>
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<td>DANSK FLY ELEKTRONIK ApS</td>
<td>B200 KING AIR</td>
<td>Installation of Garmin DUAL GPS-400, GPS-1 is linked to the autopilot system. Ref: DFE-DD-0401-PS-00 Product Specification.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td></td>
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<td>16/03/2011</td>
<td>Active</td>
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<td>DANSK FLY ELEKTRONIK ApS</td>
<td>B200 KING AIR</td>
<td>Installation of Bendix/Kings KGP 560 EGPWS system with Avidyne EX500 display. Ref: DFE-DD-0400-PS-00 Product Specification.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
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<td>16/03/2011</td>
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<td>AERODYNE REPAIR SERVICES LTD.</td>
<td>TCCA STC SH06-41</td>
<td>External MGT and Torque Indicating System Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
<td></td>
<td>02/06/2011</td>
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<td>10034211</td>
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List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<th>Rev. level</th>
<th>Foreign/NAA</th>
<th>Model</th>
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<tr>
<td>10034212</td>
<td>BOMBARDIER AEROSPACe</td>
<td></td>
<td>TCCA STC- ST03910AT</td>
<td>CL-600-2B16</td>
<td>MILAN Activation &amp; Incorporation of Aircell Components</td>
<td>EASA.MA.0237</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Transport Canada STC No. SA10-93 Issue dated 27 October 2010; CAA-CAI MMEL Supplement A320-CAI-MMEL-S-1, dated 15th March 2011. The aircraft type, model and manufacturer serial number for which this change is applicable is: Airbus A320-232 with Manufacturer Serial Number: 4166. Prior to installation of this change...</td>
<td>16/03/2011</td>
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<tr>
<td>10034237</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 1</td>
<td></td>
<td>1900D</td>
<td>Double Garmin GPS400W Installation for B-RNAV and P-RNAV and RNP/Approach (LNAV procedures, according to modification No BCA 121-06 revision 1 dated 24/04/2012; AMM Supplement No. BCA-121-05 revision 0 dated 23/12/2010; Master Drawing List No. BCA-121-09 revision 1 dated 24/04/2012; or later revisions of the above listed documents approved by EASA.</td>
<td>AFM Supplement No. BCA-121-06 revision 1 dated 24/04/2012; or later revisions of the above listed documents approved by EASA.</td>
<td>Installation limited to Beech 1900D equipped with Rockwell Collins AP/FD autopilot or FOG 65 H Flight Director and EFIS Rockwell Collins DPU 84, upgraded for GPS approach through Rockwell Collins SB 77 (523-0805196) and SBI 81 (523-0805046).</td>
<td>15/06/2012</td>
<td>Active</td>
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<tr>
<td>10034238</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td></td>
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<td>1900C, 1900D</td>
<td>BCA 120 revision 0 - 28-230 volts inverter(s) Installation on Beech 1900 Series.</td>
<td>Airplane Flight Manual Supplement BCA-120-A revision 0 dated 09 March 2011; and applicable MMEL supplements of the original product.</td>
<td>None</td>
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<td>17/03/2011</td>
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<td>10034242</td>
<td>FOKKER TECHNIEK B.V.</td>
<td>REV. 2</td>
<td></td>
<td>A320-232</td>
<td>AIRBUS A320-232 - VIP conversion Revision 1: update of applicable MMEL supplement; approved as OSD-MMEL Revision 2; Transfer of STC from Fokker Services B.V. (EASA A.S.03722) to Fokker Techniek B.V. (FAA STC ST03910AT)</td>
<td>A320-CAI-CRR-0001 Issue 3, approval date 17th March 2011; A320-CAI- AFM-S-APPROVAL issue 1, dated 15th March 2011; A320-CAI-MMEL-S- APPROVAL revision 3, dated 16th February 2017.; ICA Supplement document: SE-712 issue 2, dated 18th February 2011; or later revisions of the above listed documents approved by EASA.</td>
<td>Refer to limitations section of the approved AFM supplement A320-CAI-AFM-S- APPROVAL issue 1, dated 15th March 2011. The aircraft type, model and manufacturer serial number for which this change is applicable is: Airbus A320-232 with Manufacturer Serial Number: 1168. Prior to installation of this change...</td>
<td>05/12/2019</td>
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<td>EASA STC No.</td>
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<td>10034246</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>FAA STC SR02114NY-D</td>
<td>EC135 T2(CPD2), EC135 T2+</td>
<td>Installation of a Honeywell Model 15175 VXP System. The Honeywell Model 15175 VXP System is a Health and Usage Monitoring System that monitors vibration characteristics of the aircraft to provide data to aid aircraft maintenance. The information provided by the system is supplementary in nature and does not replace the original aircraft system or flight manual...</td>
<td>The Certification Basis for the original product remains applicable to this certificate or approval. The requirements for environmental protection and the associated noise certification or emissions levels of the original product unvarnished and remain applicable to this certificate or approval.</td>
<td>None.</td>
<td>Only applicable to Honeywell Master Drawing List ID01(152); revision B.</td>
<td>29/03/2011</td>
<td>Active</td>
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<tr>
<td>10034248</td>
<td>AERODATA AG</td>
<td>REV. 3</td>
<td>B200</td>
<td>Midlife System Upgrade AD- 28/03/2011</td>
<td>The Type and OSD Certification Bases (CB) for the original product are applicable to this certificate or approval. The requirements for environmental protection and the associated noise certification or emissions levels of the original product unvarnished and remain applicable to this certificate or approval.</td>
<td>Master Document List MDL-AVC-157, Rev. 17Flight Manual Supplement AFM-116, Rev. 8Instructions for Continued Airworthiness AMM-1300, Rev. 10No later revisions of the above listed document(s) approved in accordance with the EASA system.</td>
<td>Only applicable to B200 aircraft S/N BB-2016 and BB-2018 prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2024</td>
<td>Active</td>
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<tr>
<td>10034252</td>
<td>BINDER MOTORENBau GmbH</td>
<td>REV. 1</td>
<td>ASH 25 E</td>
<td>Conversion from hand starter to electric starter Umbau Handschalter auf Elektrostarter.</td>
<td>The Certification Basis for the original product remains applicable to this certificate or approval. The requirements for environmental protection and the associated noise certification or emissions levels of the original product unvarnished and remain applicable to this certificate or approval.</td>
<td>No.</td>
<td>21/03/2011</td>
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<td>10034322</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A330-300</td>
<td>Thai Airways A330-300 - Seat exchange and electrical adaptation. The STC installs 263 Y class seats, with associated electrical adaptations.</td>
<td>The Certification Basis for the original product and the following or alternative airworthiness requirements are applicable to this certificate or approval. TCDS EASA A-0040R F-4200-04, IFE and ISPST: The requirements for environmental protection and the associated noise certification or emissions levels of the original product unvarnished and remain applicable to this certificate or approval.</td>
<td>Supplemental Type Definition Document, ref: STD000-1059M Issue 1, dated 17 March 2011, plus later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>24/03/2011</td>
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<td>POLICE AVIATION SERVICES Ltd.</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>M002 CONFIG. P&amp;W 2068207</td>
<td>Aircraft Role Configuration</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval. CS-25 Am.27 The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>RM STC Supplement EASA 592-002 Rev A or later revisions of the above listed documents approved by EASA.</td>
<td>none</td>
<td>25/02/2011</td>
<td>Active</td>
<td>10034324</td>
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<td>GAMA ENGINEERING Ltd.</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Navigation/Communications Upgrade.</td>
<td>The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List MDSL-1208 Issue 1 Change Statement CS-1208 Issue 1 Instructions for Continued Airworthiness ICA-1208 Issue 1 Flight Manual Supplement AFMS-1208-01 Issue 1 Flight Manual Supplement AFMS-1208-02 Issue 1 Flight Manual Supplement AFMS-1208-03 Issue 1 for later revisions of the above listed documents.</td>
<td>None</td>
<td>25/03/2011</td>
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<td>DART AEROSPACE Ltd.</td>
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<td>TCCA STC 5H110-49</td>
<td>429</td>
<td>Installation of UltraPaws.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval. CS-25/27/29 Change 2 The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>MDL-D429-793 revision A dated 27 October 2010 - Master Document List (MDL-D429-793 revision B dated 8 October 2010 - Installation Instructions/ICAC-D429-793 revision C dated 18 October 2010 - Instructions for Continued Airworthiness).</td>
<td>None</td>
<td>25/03/2011</td>
<td>Active</td>
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<tr>
<td>H4 AEROSPACE (UK) Ltd</td>
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<td>FAA STC</td>
<td>5G743-400</td>
<td>Installation of a two-class seating arrangement</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The certificate/approval applies to a two-class seating arrangement.</td>
<td>Modification Summary Sheet and Declaration of Compliance MDH4A1381 Issue 1 dated 14/03/2011 or later revisions of the above listed documents approved by EASA.</td>
<td>Not Applicable</td>
<td>29/03/2011</td>
<td>Active</td>
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<tr>
<td>ARCHEION HOLDINGS, LLC.</td>
<td></td>
<td>FAA STC</td>
<td>5O7S20AT</td>
<td>Installation of an Avionics satINK Communication Management System</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval. A new area of certification is CS-25 Amendment 2 (1-&lt;25.301, 25.303, 25.305, 25.307, 25.961, 25.1351, 25.1209, 25.1351, 25.1553, 25.1).</td>
<td>Archeion Holdings LLC Master Drawing List Doc. No. &quot;ZSM-1208-001&quot; dated 25th June 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/4/EC.</td>
<td>None</td>
<td>30/03/2011</td>
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<td>airplus engineering GmbH</td>
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<td>REV. 1</td>
<td>2. 58P, 58PA, 58TC, 58TCA 1. 59A, 59A, 58</td>
<td>Dual Installation of Garmin GNS430A/NAV</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval. EASA CS-23, Amendment 2 (select to comply by the applicant, not required by 21A.101). The requirements for environmental protection and the...</td>
<td>Engineering Work Order: E36-1173-00000, Rev. 1 Flight Manual Supplement: PO-1173-00000, Rev. 1 Maintenance Manual Supplement MM-1173-00000, Rev. 1 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>18/02/2021</td>
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### List of EASA Supplemental Type Certificates

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<th>Rev. level</th>
<th>Foreign/NA A STC ref</th>
<th>Model</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>KUERZI AVIATION AG</td>
<td>10034392</td>
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<td></td>
<td>MYSTERE-FALCON 50</td>
<td>FMS upgraded module of the existing UAS-1/4 to UAS-1/6V. Installation of a second FMS Upgrade of the existing Collins VFR-31 1/2 and DME-41 1/2 to VFR-2000m DME-42 so that ARINC-429 can be used to interface with the dual FMS.</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>Master Data/Document/Drawing List doc. N° MDL-2010/449 Rev. A dated 03/01/2011/ Master Packag List doc. N° MRL-2010/34519 Rev. N/C dated 03/03/2011/ Aircraft Flight Manual Supplement doc. N° FMS-2010/34419 Rev 0 dated 28 March 2011/Instructions for Continued Airworthiness, doc. N° ICA-2010/34419 Rev N/C dated 10 March 2011.</td>
<td>Restricted to MSG 5/6 183</td>
<td>31/03/2011</td>
<td>Active</td>
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<td>10034408</td>
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<td>AS 350 B3 &amp; AS 350 B3</td>
<td>Installation of Electromagnetic Ground Ice Thickness Sensor</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. This certification/approval involves a change to the certified noise and emissions levels.</td>
<td>FAA approved Drawing No. 20002-0000, Rev. A, dated 29 Apr 2006, FAA approved Aircraft Flight Manual Supplement APN20002-1, Rev. II, dated 06-Juli-2006, is required for Baron 58, 58A, and 58B airplanes, serial numbers TH-1399, TH-1396 and forth subsequent instructions for Continued Airworthiness, Doc. No. 20002-30.d...</td>
<td>Only aircraft equipped with a Teledyne Continental IO-550-C engine is eligible for this STC</td>
<td>31/03/2011</td>
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<td>HARTZELL PROPELLER INC.</td>
<td>10034304</td>
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<td>58, 58A, 58B</td>
<td>Installation of Hartzell PHC-25YF-2UF/ FFC7391D(B, K) propellers</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>FAA approved Document No. 2587-02-323, rev. 0 dated 09/01/2011</td>
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<td>01/04/2011</td>
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<td>AVIATION ENGINEERING AS</td>
<td>10034408</td>
<td></td>
<td></td>
<td>AS Master Drawing List - 0 dtd 21.02.2011 or later revisions of the above listed drawings approved by EASA/Aviation Engineering AS Master Drawing</td>
<td>Installation of Bendix/King EFI-890R as specified in Aviation Engineering AS COMPLIANCE CHECK LIST doc. No. 2587-02-323, rev. 0 dated 09/01/2011.</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>Aviation Engineering AS Flight Manual Supplement AS350B2, B3 PostModification Eurocopter SBT 25.00.62 - doc. No. 2587-02-333, rev. 0 dated 21.01.2011 or later revisions of the above listed drawings approved by EASA/Aviation Engineering AS Master Drawing List - doc. No. 2587-02-324, rev. 0 dated 14.01.2011</td>
<td>AS350B2 and AS350B3 post mod Eurocopter SBT 25.00.62</td>
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<td>SHAIDN AVIONICS</td>
<td>10034421</td>
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<td>250-C47B, 250-C47M</td>
<td>Installation of Shadin fuel flow transducer in the fuel line between the fuel control unit (FCU) and the fuel injector. Additional instructions for EASA Approval Model List (JAML) No. 5308421 of the Rolls-Royce Corporation engine/mission 250-C300/A1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>-FAA STC SE578GL - Application letter “Request for EASA STC Major Change Level 1” from Mr. Brad Jacobson (Shadin Avionics) to Ms. Khalea Hosmy (FAA), dated May 10th, 2017</td>
<td>None</td>
<td>12/06/2017</td>
<td>Active</td>
<td>10034421</td>
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<td>MT-PROPELLER ENTWICKLUNG Gmbh</td>
<td>10034434</td>
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<td>PA-46-310P</td>
<td>1) Installation of MTV-14-D/190-17 on PA-46-310P (installation of MTV-14-D/195-30s on PA-46-310P)</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>For installation 1 and 2:- &quot;Anhang zum Flughandbuch Nr. E-121&quot; Issue 30.09.1996; &quot;Umleitung Weisung Nr. E-511&quot; Issue 30.09.1996; For installation 3:- &quot;Anhang zum Flughandbuch Nr. E-666&quot;, Issue 22.09.1997; &quot;Umrouting Weisung Nr. E-666&quot;, Issue 22.09.1997</td>
<td>None</td>
<td>25/05/2011</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<th>Rev. level</th>
<th>Foreign/NAAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<td>10034435</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 1</td>
<td>LBA STC (EMD) 20A626 ISSUE 4</td>
<td>PA-46-360P PA-46R-350T</td>
<td>1) Installation of MTV-14-B/196-172 Installation of MTV-14-B/195-350 For installation 3 optional reduction of take-off propeller speed to 2400 rpm is included.</td>
<td>The Certification Basis for the original product and the followon/bidirectional or alternative airworthiness requirements are applicable to this certificate for approval. FAR 23 including Amendment 34 (not required by 21A.101. elected to comply by the applicant). This certificate approval involves a change to...</td>
<td>None</td>
<td>01/04/2011</td>
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<td>10034437</td>
<td>JET AVIATION AG. BASEL</td>
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<td>In accordance with the EASA TCDS The Certification Basis for the original product and the followon/bidirectional or alternative airworthiness requirements are applicable to this certificate for approval. - CRI H-01 “Enhanced Airworthiness Programme for Aeroplane Systems - ICAon EWIS” The requirements for envr...</td>
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<td>10034446</td>
<td>TENENCIA Ltd.</td>
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<td>Cockpit Door Surveillance System Certification Basis as per EASA TCDS MA A.080 issue 3, 28thOctober 2010/10a. The applicant elected to comply with JAR 25 at change 15 for all models. In addition, CRI H1 issue 2 dated 23/02/2011 on &quot;Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS&quot; applies. The requirements for envr...</td>
<td>None</td>
<td>05/04/2011</td>
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<td>10034447</td>
<td>AEROSPACE DESIGN AND CERTIFICATION LTD. T/A AERODAC</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>V737-700, -800, -900</td>
<td>Cabin certification for aircraft use in Commercial operations</td>
<td>The Certification Basis for the original product is unchanged and remains applicable to this certificate. The requirements for environmental protection and the associated telemetry noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>None</td>
<td>12/04/2011</td>
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<td>10034457</td>
<td>SHADR AVIONICS</td>
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<td>Installation of Digital Fuel Flow Meter.</td>
<td>The Certification Basis for the original product remains applicable to this certificate for approval. The requirements for environmental protection and the associated telemetry noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>None</td>
<td>08/04/2011</td>
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<td>TRUMPINS INSULATION SYSTEMS, LLC.</td>
<td>REV. 1</td>
<td>FAA STC ST2013LA</td>
<td>B767-200</td>
<td>Insulation Blanket Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated telemetry noise and/ or emissions levels of the original product are unchanged and remain applicable to this approval.</td>
<td>None</td>
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<td>CAIRNS SHOCK TRAUMA AIR RESCUE (CALSTAR)</td>
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<td>Engine Upgrade Removal of two (2) Rolls-Royce engine models 250-C28C and installing two (2) Rolls-Royce engine models 250-C30P in accordance with FAA STC SR02160LA, dated March 11, 2011.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- CALSTAR FAA approved Rotorcraft Flight Manual Supplement CAL204 Revision 1R, dated July 30, 2008 or later revisions of the above listed documents approved by EASA - CALSTAR Master Drawing List CAL201, Revision B, dated May 19, 2008. Installation of (2) Rolls-Royce 250-C30P Engines, Calstar 203-IC.</td>
<td>None.</td>
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### List of EASA Supplemental Type Certificates

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<td>10034622</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>BEECH 1900D</td>
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<td>Beech 1900D - Full VIP Cabin Installation for Cessna 114, 114A, 114B, 112TC, 112TCA</td>
<td>Dual installation of Garmin GNS430(A)/GNS530(A)/GATS5</td>
<td>Engineering Work Order: E39-1162-00030, Rev. 1</td>
<td>19/04/2011</td>
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<td>10034623</td>
<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td>114, 114A, 114B, 112TC, 112TC</td>
<td>Sky Diving Kit for Cessna 114, 114A, 114B, 114TC</td>
<td>Engineering Work Order: E39-1162-00030, Rev. 1</td>
<td>None</td>
<td>16/02/2021</td>
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<td>AIROTEC - T UML GmbH &amp; CO. KG</td>
<td>REV. 2</td>
<td>206H, T206H</td>
<td>Sky Diving Kit for Cessna 206H</td>
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<td>None</td>
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<td>10034631</td>
<td>AEROTEC - T UML GmbH &amp; CO. KG</td>
<td>REV. 3</td>
<td>P2006T</td>
<td>Sky Diving Kit for Cessna 206H</td>
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<td>BRISTOW HELICOPTERS LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>S-76C</td>
<td>Qualification of the Garmin GNS-530 Series GPS and/or Flight Manual Supplement for En-route Navigation and Non-precision approaches</td>
<td>None</td>
<td>29/03/2012</td>
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<td>10034641</td>
<td>2 EXCEL DESIGN LIMITED</td>
<td>REV. 5</td>
<td>CAA UK RESPONSIBILITY</td>
<td>PA-31</td>
<td>Introduction of Chin Mount and re-located Pilot tubes</td>
<td>None</td>
<td>25/06/2017</td>
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<td>LEITH &amp; ASSOCIATES, LLC</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>ST01857E</td>
<td>757-200</td>
<td>Maximum zero fuel weight increase</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Drawing List Leth &amp; Associates document No. OL0236. Revision NC(++)(++) &gt; dated 19 June 2008. - Airplane Flight Manual Supplement, Leth &amp; Associates document No.OL0224(++)(++) - Revision NC, dated 15 June 2008, or later revisions of the above listed documents approved by EASA in accordance with EASA...</td>
<td>20/04/2015</td>
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<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.L.U.</td>
<td>REV. 1</td>
<td>AS 365N3</td>
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<td>Adoption of a FLIR ULTRA 4000 system installation provisions for a FLIR ULTRA FORCE II. This modification is a minor change of the STC FLIRULTRA 4000 - 365N3 REF. AHSE-50300 of ASTEC HELICOPTER SERVICES AS; keeping the supporting structure and the electrical provisions for FLIRULTRA 4000. Rev. 1 Trans...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Anworthiness Document CMA044-AAA E1 ROC/Engineering BulletinCMA634-B1 E1 ROC/Maintenance and Installation/Uninstallation/Instructions CMA034-MID E-DDrawings ZW-4503-01 A HELICSA, ZW-2010-01 H, ZW-2009-01 H &amp; VOR FOR LEOSAFM Supplement CMA044-1&amp;2 AFMS E1 R0r later revisions of the above listed documents...</td>
<td>FLIR ULTRA 4000 must be previously fitted according to the STC FLIRULTRA4000 - 365N3 REF. AHSE-50300 of ASTEC HELICOPTER SERVICES AS Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed change and/or...</td>
<td>01/02/2019</td>
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<td>JET AVIATION AG, BASEL</td>
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<td>MF900 (EXCEPTED FFROC)</td>
<td>NZ2000 6.1 upgrade with CD820 CDU replacement F-Holding dealer triple Honeywell NZ2000 Flight Management System (augmented as follows: - NCU 3400 FM S&amp;i version NZ52 later is replaced with version NZS1.1 CD800/ CD810/1&amp;2 CDU replaced with CD820 No change in sensors and capabilities.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List Doc. No. BFF0034623-001 Rev. B dated 01.02.2011Instructions for Continued Airworthiness Doc. No. BFF0034623-150-01 dated 20.12.2010/Airplane Flight Manual Supplement Doc. No. BFF0034623-201-01 Rev. 0 dated 17.03.2011or later revisions of the above listed documents approved by EASA.</td>
<td>As per AFMS Doc. No. BFF0034623-201-01 Rev. IR dated 17.03.2011.</td>
<td>18/04/2011</td>
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<td>HELICOPTER AIR TRANSPORT GmbH</td>
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<td>EC135 T2</td>
<td>Dual GNS430V Installation in Autopilot helicopter</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>AFMS 34-002, Revision 0 for later revisions of the above listed documents approved by EASAEngineering Order EO 34-002, Revision 0, including referenced drawings and Appendix A and Bininstallation Manual EO 34-002/MM 34-002 Maintenance Instructions CMA034-AAA E1 ROC/Engineering Bulletin CMA634-B1 E1 ROC/Maintenance and Installation/Uninstallation/Instructions CMA034-MID E-DDrawings ZW-4503-01 A HELICSA, ZW-2010-01 H, ZW-2009-01 H &amp; VOR FOR LEOSAFM Supplement CMA044-1&amp;2 AFMS E1 R0r later revisions of the above listed documents...</td>
<td>The installation is limited to Aircraft Serial Numbers 0220 and 0221</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 1</td>
<td>C-212-ED (C212-200) C-212-CC (C-212-200)</td>
<td>MOD 0021. ESGPS installation This revision extends the applicability to add C-212-CC (C-212-200) MSN139.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>MASTER DATALOG AE0921MDL Rev. 10 Dated 22. Sept. 2019. For MSN 387 SUPPLEMENT AU MANUEL DE MAINTENANCE AE0921 IMRN REV 0 dated 09. Oct. 2009 SUPPLEMENT AU MANUEL DE VOL AE0921 SHM REV 0 dated 20. Apr. 2011 For MSN 130 AIRCRAFT FLIGHT MANUAL SUPPLEMENT AE0921AFMS 3 REV 0 dated 22. July 2016 INSTRU...</td>
<td>This STC is limited to MSN 387 and MSN 139 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/10/2016</td>
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<td>EASA STC No.</td>
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<td>10034669</td>
<td>RUECKER AEROSPACE GmbH</td>
<td>328-300</td>
<td></td>
<td>Modification to certificated LOPA</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise levels or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Declaration of Compliance of Ruecker Aerospace Entwicklungsabteilung/EASA 211.295 dated 19.04.2011</td>
<td>none</td>
<td>19/04/2011</td>
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<td>10034675</td>
<td>AEROTEC &amp; CONCEPT</td>
<td></td>
<td>C90A</td>
<td>Installation d'une antenne SATCOM iridium - dossier aerotec1082rev1</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CS 23.303, 305, 307, 365, 571, 573, 603, 605, 607, 609. The certificated noise levels or emissions levels of the original product are unchanged.</td>
<td>none</td>
<td>none</td>
<td>19/04/2011</td>
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<td>10034675</td>
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<td>10034682</td>
<td>OGMA, INDUSTRIA AERONAUTICA DEPORTUGAL, S.A.</td>
<td></td>
<td>EMB-135BJ</td>
<td>Change Number 1086.07 - &quot;Installation of coffee maker &amp; associated water tank&quot;</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise levels or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>- Change approval sheet: CAS- A1.50C-1086.07-25.00-001 dated 14 Apr 2011 - Change Dossier: CD- A1.50C-1086.07-25.00-001 Rev A dated 14 Apr 2011 - Aircraft Maintenance Manual Supplement: MS- A1.50C-1086.07-00.00-003 initial issue, dated 23 Nov 2010 or later revisions of the above listed documents approved by EAS...</td>
<td>none</td>
<td>19/04/2011</td>
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<td>10034690</td>
<td>AIRBUS DEFENCE AND SPACE S.A.</td>
<td>REV. 20</td>
<td>A330-243</td>
<td>A330 MRTT FSTA Modification STC/Rev. 1: Deletion of a previous limitation on the operation of the lift system - Review of the role change capability. Therefore, it is possible for an aircraft to move from the military register (INTA Technical Certificate configuration) to the civil register...</td>
<td>The Type and OSD Certification Bases for the original product are amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25.1853 Initial Issue for affected areas by change DCV1088-EThis certificate/approval involves a change wil...</td>
<td>- &quot;Conversion Definition A330MRTT FISTA&quot; document ref. ME-FU- DT-110145, issue A - ALS Repository /A330 FISTA EASA STC1 Airworthiness Limitation Section, document DT-FU- C00-0019, issueA - ALS Part 1 Supplement /A330 FISTA Safe-Life Airworthiness Limitation Item Supplement for EASA STC1, document DT-FU-C0...</td>
<td>none</td>
<td>17/01/2024</td>
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<td>10034699</td>
<td>AERONAUTICAL ACCESSORIES, INC</td>
<td></td>
<td>429</td>
<td>Installation of Health and Usage Monitoring System (HUMS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise levels or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>- AA-09040 revision 0 dated 9 November 2010 - Flight Manual Supplementor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 200404/AF (or subsequent revisions of this decision) - AA-09033 revision D dated 26 October 2010 - Master Document List-AA-09041 rev...</td>
<td>none</td>
<td>20/04/2011</td>
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<td>STC Holders</td>
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<td>ACS-NAV Ltd</td>
<td>10034703</td>
<td>REV. 1</td>
<td>TC2A STC: SA07-40</td>
<td>PC-12/46; PC-12/47; PC-12/47E</td>
<td>Installation Utility Door Existing Cargo door modified to incorporate a utility door (900 mm high, 635 mm wide)</td>
<td>FAR 23 The Certification Basis for the original product remains applicable - no certificate approval</td>
<td>Flight manual supplement Doc. No. F195000; Modification data summary Doc. No. F195000; Maintenance manual supplement Doc. No. E55070-MMS, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/C (or subsequent revisions thereof).</td>
<td>none</td>
<td>28/11/2012</td>
<td>Active</td>
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<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>10034706</td>
<td>REV. 1</td>
<td>AS 350 B, AS 350 B1, AS 350 B2, AS 350 B3</td>
<td></td>
<td>Installation of Air Ambulance Equipment (removable stretcher)</td>
<td>The Certification Basis (CB) for the original product remains applicable for the following additional or alternative airworthiness requirements are applicable to this certificate/approval: FAR 27 Fire protection; FAR 27 Fire protection</td>
<td>Flight manual supplement Doc. No. AS07-40 Failure Data Summary Doc. No. AS07-40; Maintenance manual supplement Doc. No. E55070-MMS, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/C (or subsequent revisions thereof).</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/07/2018</td>
<td>Active</td>
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<tr>
<td>2 EXCEL DESIGN LIMITED</td>
<td>10034714</td>
<td>REV. 4</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>PA-31</td>
<td>Introduction of enlarged window and sensor turret provisions</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List S21.MDL-0914 Issue 7; Flight Manual Supplement S21.FMS-0077 Issue 4; Compliance Record S21.COMP-0914 Issue 4 Instructions for Continuing Airworthiness S21.TEC-0374 Issue 2;</td>
<td>With a sensor turret with diameter greater than 15.5” installed, right-to-left or forecast icing conditions is prohibited. Limitations as defined in AFMS S21.FMS-0077 Issue 4. The installation and subsequent operation of a sensor turret requires separate approval prior to installation of this mod.</td>
<td>25/08/2017</td>
<td>Active</td>
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<tr>
<td>2 EXCEL DESIGN LIMITED</td>
<td>10034734</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>PA-31</td>
<td>Introduction of a Belly Port</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List S21.MDL-0915 Issue 3 dated 13/11/2012. Flight Manual Supplement S21.FMS-0090 Issue 3 dated 31/10/2012; Compliance Record S21.COMP-0915 Issue 2 dated 13/11/2012; Instructions for Continuing Airworthiness S21.TEC-0311 Issue 1 dated 20/04/2011 or later revisions of the above listed.</td>
<td>Not to be flown in icing conditions with the belly pod installed. Limitations as defined in AFMS S21.FMS-0090 Issue 2 dated 31/10/2012. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change.</td>
<td>25/08/2017</td>
<td>Active</td>
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<td>ARMSTRONG AEROSPACE Inc.</td>
<td>10034747</td>
<td></td>
<td>FAA STC ST0293CH-D</td>
<td>B737-800</td>
<td>Validation of FAA STC ST0293CH-D PANASONIC DMPS - In-flight Entertainment System installation</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: Certification Basis of EASA TCDS (MA A 120 and CRI H 01 Issue 01 on EWIS)&lt;(&gt;) CRI F 01 Issue 01 on IFE The requirements for environmental protection and th...</td>
<td>- Master Data List. Ref: DC104-8084-00 Rev. B, dated 22 April 2011. Airplane Flight Manual Supplement, Ref: DC303-8084-00 dated 15 April 2011; Instruction for Continued Airworthiness Supplement. Ref: DC113-8084-00 Rev. A, dated 15 April 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>27/04/2011</td>
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<td>STC Holder</td>
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<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td>REV. 3</td>
<td>AS 350 B</td>
<td>Installation of video transmission equipment PP-013Rev. 01, New cabin rack 1/2 (PN PP-10-A051 added)Rev. 02 Renumbering of elements and enhancement of installation instructionsRev. 03 correction of typo concerning Rotorcraft Flight Manual Supplement</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List PP-013-MLD-200-00 Issue 0 dated 02 June 2021/ Instruction for Continued Airworthiness PP-013-ICD-200-00 issue 0 dated 02 June 2021/Rotorcraft Flight Manual Supplement PP-013-AMF-03 issue 3 dated 30 March 2022 or later revisions of the above listed document(s) approved/accredits the EASA STC.</td>
<td>3/06/2021</td>
<td>Active</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>A380-841, A380-842</td>
<td>DLH Stretcher Installation</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approvalCRI D-51 Special Condition on Stretcher Installations The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Definition Document, STD225-10/23M, Issue 1, Rev. 0, 15/04/2011/Declaration of Compliance No. 3151, 18/04/2011 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>29/04/2011</td>
<td>Active</td>
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<td>AERGO SOLUTIONS INC</td>
<td>REV. 2</td>
<td>PC-12, PC-12/45 PC-12/47, PC-12/47E</td>
<td>PC-12 EASA STC 10034806 Rev. 1, Canadian Holder transfer to Aergo/Due to a Canadian STC Holder transfer (from ACS-NAI Ltd. to Aergo Solutions Inc.), the STC has to be updated to Revision 2. In addition there was a clarification on the configuration that needed a new &quot;editorial&quot; limitation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>GS04000 Rev C Master Data List GS04000 Rev B AFMS &quot;Installation Feather Inhibit System&quot; GS04071 Rev C Maintenance Manual Supplement &quot;Installation Feather Inhibit System&quot; or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>20/01/2016</td>
<td>Active</td>
<td>10034806</td>
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<tr>
<td>GAMA AVIATION (ENGINEERING) LIMITED</td>
<td>REV. 1</td>
<td>A350 B, A350 B1, A350 B2 AS 350 B, AS 350 B3, AS 350 BA, AS 350 D</td>
<td>Flight Deck Manual Stowage Box TCCA</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>All of the documents detailed in GAMA Aviation Master Document List (MDL) MDL-1224 Issue 4.</td>
<td>None</td>
<td>17/03/2015</td>
<td>Active</td>
<td>10034814</td>
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<td>AMTECH AERONAUTICAL LIMITED</td>
<td>TC78A60-70</td>
<td>Installation of a Rosemount Aerospace Angle of Attack Sensor.</td>
<td>EASA JMA.023 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Rosemount Aerospace Inc. (RMTAERO) Master Data List D06404350, Revision A or later approved revision(=&gt;) or Rosemount Aerospace Inc. Maintenance Manual Supplement and Instructions for Continued Airworthiness D06404348, Revision A or later approved revisions(=&gt;) or later revisions of the above listed documents.</td>
<td>None</td>
<td>05/06/2011</td>
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<td>ACS-NAI Ltd.</td>
<td>TC78A60-70 ISSUE 2</td>
<td>FAR 25 as per EASA JMA.191, Iss. 4 for DHC-8-102/200/300 Series AIR-2S-28 as per EASA JMA.191, Iss. 4 for DHC-8-400 Series The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation must be in accordance with ACS-NAI Ltd. TC78A approved Modification Data Summary B 78600 Rev. 00. Maintenance must be in accordance with Instructions for Continued Airworthiness MMS B 78671(=&gt;) or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>06/06/2011</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<td>10034876</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 1</td>
<td>10034881</td>
<td>C-212-2D (C-212-300) C-212-CC (C-212-200)</td>
<td>MOD 006: Installation of a honeycombed CA6157 A TCS31 system and A/C ModExi transponder MST571X. This revision extends the applicability to add C-212-CC (C-212-200) MSN139.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this STC. Approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>MASTER DATA LIST AE0008MLD Rev. 11 Dated 22 Sept. 2011 For MSN 139.</td>
<td>-INSTRUCTIONS FOR AIRWORTHINESS MAINTAIN: AE0006PMN REV 3 dated 04 Apr. 2011. FLIGHT MANUAL SUPPLEMENT: AE0008SNM REV 2 dated 17 Mar. 2011 For MSN 139. AIRCRAFT FLIGHT MANUAL SUPPLEMENT: AE0008AFMS_3 REV 0 dated 22 Jul. 2011: INSTRUC...</td>
<td>26/10/2016</td>
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<td>10034881</td>
<td>AERO DYNAMIX, Inc.</td>
<td>FAA STC</td>
<td>S-76C</td>
<td>Installation of Night Vision Imaging System (NVIS) / Night Vision Goggle (NVG) compatible lighting system.</td>
<td>The Certification Basis for the original product remains applicable to this STC. Approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List (MDL), Aero Dynamix, Inc. drawing number MDL4783-01, revision D, dated 8 December 2010 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). Rotorcraft Flight Manual Supplement (RFMS) for list of equipment mandatory for Night Vision Imaging System (NVIS) certification.</td>
<td>Refer to applicable Rotorcraft Flight Manual Supplement (RFMS) for list of equipment mandatory for Night Vision Imaging System (NVIS) certification.</td>
<td>10/05/2011</td>
<td>Active</td>
<td>10034881</td>
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<td>10034885</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC</td>
<td>C90GT, E90, H90 B90, C90, C90A, C90GT 65-A60-2, 65-A60-3, 65-A60-4, 65-90, 65-A60, 65-A60-1</td>
<td>Installation of Vortex Generators on the wings and tail surfaces, and retakes on the engine nacelles.</td>
<td>The Certification Basis for the original product remains applicable to this STC. Approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Micro Aerodynamics Drawing Package MA2236, Revision IR, dated October 06, 2010, and Installation Manual MA2237, Revision IR, dated December 01, 2010. Instructions for Continued Airworthiness in accordance with Micro Aerodynamics Document MA0666, Revision A or later revision(s).</td>
<td>If more than five vortex generators, or either nacelle fail, the aircraft is damaged. The aircraft is no longer airworthy.</td>
<td>10/05/2011</td>
<td>Active</td>
<td>10034885</td>
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<td>10034891</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>FAA STC</td>
<td>A340-313</td>
<td>Installation of EFB class II/This STC installs the Rosemount Aerospace 9700EZC-3 Laptop Docking Station (System provision)</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this STC. Approval. EASA TCDS EASA.A.015CRF -F2005-002: Installed Provisions for EFB Class 2 Provisions. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, ref: AE1098MLD, dated 09 May 2011. Aircraft Flight Manual Supplement, ref: AE1098SNM, dated 09 February 2011. AE1098PMN, dated 18 March 2011. Instructions for Continued Airworthiness, ref: AE1098PMN, dated 17 March 2011 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>10/05/2011</td>
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<td>10034892</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>FAA STC</td>
<td>A330-200</td>
<td>A330-200 Cabin retrofit ETD01</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this STC. Approval. EASA TCDS A.04Issue 25: Special Condition E1014 Issue 4 (Installable SeatBall) The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental Type Definition Document, ref: STD000-1026M Issue 1 on later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>10/05/2011</td>
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<td>DASSAULT FALCON SERVICE</td>
<td>10034903</td>
<td>F2000, F2000EX</td>
<td>The Certification Basis for the original product remains applicable to the aircraft certified to the original product. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>DFS Change Approval Sheet DFS1244-CA5 Issue A dated 10.05.2011 or later</td>
<td>Modification applicable to F2000 &amp; F2000 EX aircraft with &gt;1400W available on A4 bus (version 1) or &gt;1280W (version 2)</td>
<td>11/05/2011</td>
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<td>BOMBARDIER INC.</td>
<td>10034902</td>
<td>TCCA STC C-LSA11-024/D</td>
<td>Installation, Antenna Provisions EASA M.A.009!The Certification Basis for the original product remains applicable to the aircraft certified to the original product. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC No. C-LSA11-024/D Issue 1 dated 03 May 2011 ACS-NAI Ltd. Modification Data Summary G731000 Rev NC or later</td>
<td>None</td>
<td>11/05/2011</td>
<td>Active</td>
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<tr>
<td>LUFTHANSA TECHNIC AG</td>
<td>10034913</td>
<td>747-400 Upgrade of first class cabin replacing 16 first class seats by 16 first class seats and installation of 8 single beds and half size partitions in accordance with STD225-013M Issue 01, rev 00(+,-)</td>
<td>The Certification Basis for the original product remains applicable to the aircraft certified to the original product. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>CP25-10_138 LH FC Update Revision Declaration of Compliance ADNC 3162 d.d. 11-05-2011</td>
<td>None</td>
<td>11/05/2011</td>
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<td>SAFRAN ELECTRONICS &amp; DEFENSE (SS UNIT - HELICOPTERS DATAMANAGEMENT NT SOLUTIONS &amp; SERVICES)</td>
<td>10034914</td>
<td>EC135 T3, EC135 T2, EC135 T1, EC135 P3, EC135 P2, EC135 P1</td>
<td>Installation of Flight Data Monitoring System HELICOM V2 NUMI and HELICOM V2 NUM+Rev 1. - Introduction of reference to Special Condition for Lithium Battery Installations. which was inadvertently omitted during the initial certification.Revision 2: - Extension of the applicability of the STC to the E...</td>
<td>Applicable documentation for installation of HELICOM V2 NUMI and HELICOM V2 NUM+Rev 1. or later approved revisions.</td>
<td>Prior to installation of this design change, it shall be determined that the design change shall not introduce any adverse effects upon the airworthiness of the aircraft.</td>
<td>11/05/2011</td>
<td>Active</td>
<td>10034914</td>
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<tr>
<td>BEECHCRAFT CORPORATION</td>
<td>10034918</td>
<td>G1000 Integrated Avionics System update with options including Synthetic Vision and Pathways and GTS 820 Traffic Advisory SystemEASA Approval of SBAA based LNAV/ILS/NAV and LPV Functionality Revision 1: Transfer from Garmin International Inc. to Beechcraft Corporation</td>
<td>The Certification Basis for the original product remains applicable to the aircraft certified to the original product. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Garmin Master Drawing List (MDL) 005-006G-18, Revision 1, dated July 21, 2010. OperationIn accordance with Garmin FAA Approved Flight Manual Supplement 156-11258-02, Revision 1, dated July 21, 2010, and EASA Specific Flight Manual Supplement ref 156-11258-02 Revision 1...</td>
<td>Aircraft modified with this STC must be maintained in accordance with the FAA approved instructions for continued airworthiness. Other previously installed design change and any other modifications must be determined by the AOC to be compatible with the design change.</td>
<td>11/05/2011</td>
<td>Active</td>
<td>10034918</td>
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<td>EASA STC no.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10034925</td>
<td>ROEDER PRAE2ISION GMBH</td>
<td>REV. 2</td>
<td>T303</td>
<td>Installation of 1. Dual GPS/NAV/COMM Garmin GNHS400W/COM/NAV/IGPS, EHSI, Sandel SN3308 with Directional Gyro Century N3200AS, Various other NAV/COMM related Avionic Components</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval</td>
<td>Engineering Order: C303- NY-07165-05, Rev. 1/Flight Manual Supplements C303- NY-07165-06a, Rev.1C303-NY-07165-06b, Rev.1C303-NY-07165-06c, Rev. 1C303-NY-07165-06d, Rev. 1C303-NY-07165-06e, Rev.1C303-NY-07165-06f, Rev.1-Maintenance Procedures C303- NY-07165-07, Rev. 1/for later revisions of the above listed documents/approved/approved/accepted by the EASA system.</td>
<td>This STC consists of three blocks (see Equipment List) which may be installed independently from each other. The components of the third block (see Equipment List) may also be installed independently. The third block of this STC includes the installation of Honeywell (Bendix/King) 4DF-KR-87 with indicator K227.</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10034925</td>
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<tr>
<td>10034927</td>
<td>FLINT AERO, INC.</td>
<td>FAA STC</td>
<td>SA252WHE</td>
<td>Installation of twelve (12) gallon auxiliary fuel tank in each wing in acc. with FAA STC SA252WHE.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Installation Instruction Flint Aero, Inc. Doc. No. FA180, Rev. L, dated September 4, 2007/Operational Limitation: see placards in acc. with Flint Aero, Inc. Doc. No. FA180, Rev. L, dated September 4, 2007/Maintenance Instruction in acc. with Flint Aero, Inc. Doc. No. FA180, Rev. L, dated September 4, 2007</td>
<td>The limitations and conditions of FAA TCDS 3A24 apply, except as outlined below: a) Model 185 through A185F S/N 18502262 - 89 gal. (85 gal. useable), two 32.5 gal. main tanks in wings at (+48) and two 12 gal. (11.5 gal. usable) auxiliary tanks in wings at (+50) b) Model A185F S/N 18502263 through 1850238, e...</td>
<td>13/05/2011</td>
<td>Active</td>
<td>10034927</td>
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<tr>
<td>10034990</td>
<td>SABENA TECHNICS BGC</td>
<td>FAA STC</td>
<td>AS 369N3</td>
<td>Installation of structural provisions (floor fittings) for cabin layout/modification in accordance with AEROCONSEL Installation Service Bulletins 2018-01, A-3101.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Aeroconsel Master Data List (MDL) 2018-01-A-2103-R00/Aeroconsel Installation Service Bulletin 2018-01-A-3101-R00/Aeroconsel Instruction for Continued Airworthiness (CAA)Q-2018-A-2211-R00</td>
<td>None</td>
<td>16/05/2011</td>
<td>Active</td>
<td>10034990</td>
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<tr>
<td>10034980</td>
<td>AVIATION SUPPORT GmbH</td>
<td>REV. 1</td>
<td>SA 365 N</td>
<td>Title: SA 365 EMS Equipment Description: Installation of an EMS Equipment fixed at the waist and cargo attachment points of the passenger compartment. The System is installed at the location of the first and second seatrow. The aft seat row remains for the health personnel. The medical equipment is installed...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Flight manual Supplement, EMS Equipment, SA365N, ASEE-91102 - EMS-91102, Revision 0-3 dated 09 May 2011 (&lt;1);&lt;1&gt;;&lt;1&gt;or later revisions of the above listed documents/approved by EASA/Lade der gültigen Dokumente (Master Document List), ASEE-91102 - EMS Equipment, SA365N, MDL-ASE-91102, Revision 0-2 dated</td>
<td>This STC is applicable for the Serial Number 6100. The use of the electrical components of the medical equipment is limited to VFR Operations. The use of a host is not possible if the EMS Equipment is installed.</td>
<td>24/05/2011</td>
<td>Active</td>
<td>10034980</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>FAA STD</td>
<td>Foreign/NAA STC No.</td>
<td>Model List</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>BEECHCRAFT CORPORATION</td>
<td>REV.1</td>
<td>FAA STD</td>
<td>SA15S46V1-D</td>
<td>G58</td>
<td>Installation in accordance with Garmin Master Drawing List (MDL) 005-00606-01, Revision 1, dated June 4, 2019. Operation in accordance with Garmin FAA Approved Flight Manual Supplement 190-01180-01, Revision 1, dated June 4, 2010, and EASASpecific Flight Manual Supplement ref 190-01180-05 Revision 1 or...</td>
<td>Aircraft modified with this STC must be maintained in accordance with the instructions for continued airworthiness (CAI) listed in the Garmin MDL identified above. Prior to installation of this modification it must be determined that the installation does not introduce any adverse effect upon the airworthiness of the product.</td>
<td>12/06/2015</td>
<td>Active</td>
<td>10034950</td>
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<tr>
<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U.</td>
<td>REV. 1</td>
<td>SA 330 J</td>
<td>Modification of transponder wiring installation to comply with EASA AD No 2010-0067Rev. 1: Transfer of STC</td>
<td>Master document list INAER.M1101034-LMD, Revision 0-Engineering bulletin INAER.M1101034-88 0, Revision 0-Declaration of compliance INAER.M1101034/DIC-1 dated 17 May 2011 or later revisions of the above listed documents approved/accepted and the EASA system.</td>
<td>Prior to installation of this modification and any other previously installed change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/02/2019</td>
<td>Active</td>
<td>10035000</td>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>REV.1</td>
<td>FAA STD NO. SR0026RC-D</td>
<td>EC135 T2(CPDS), EC135 T1(CDS), EC135 P2+ (CPDS), EC135 P1(CDS), EC135 P1(CPDS), Installation of a Belly and Fin Strobe lights in accordance with AEC Drawing List 13SA42-8018.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/interconnection system of the original product remains applicable and remain applicable to this certificate/ approval.</td>
<td>- AEC Drawing List 13SA42-8018, Revision A, dated 08/2001 Instructions for Continued airworthiness 13SM88-014, dated 07/31/10 or later revisions of the above listed documents.</td>
<td>None.</td>
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<td>23/06/2011</td>
<td>Active</td>
<td>10035015</td>
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<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>ST106854C-D</td>
<td>Installation of console table in the main cabin area of the aircraft side ledge. This table design will typically be located between benches and is not designed to be installed in front of the emergency escape hatch. Validation of FAA STC N° ST106854C-D</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>DJJ Certification Data List doc. N° F00-01010 Revision K dated April 2, 2011. Instructions for Continued Airworthiness.</td>
<td>none</td>
<td>09/05/2011</td>
<td>Active</td>
<td>10020017</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>ST106575C-D</td>
<td>Installation of assist handle validation of FAA STC N° ST106575C-D</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>DJJ Certification Data List doc. N° F00-01013 Revision J dated April 7, 2011. Instructions for Continued Airworthiness.</td>
<td>none</td>
<td>09/05/2011</td>
<td>Active</td>
<td>10020017</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>ST106854C-D</td>
<td>Installation of provision for the use of a Class 2 Dual Gooseneck Electronic Flight Bag (EFB) Smart Display System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>TEN 674 issue 38 – Modification Report (Master Data List) dated 17.12.2020</td>
<td>None</td>
<td>11/12/2020</td>
<td>Active</td>
<td>10020047</td>
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<tr>
<td>AIRLIFT A/S</td>
<td>REV. 2</td>
<td>AS365N3 AS 365 N3</td>
<td>Initial revision: Installation of NVIS Lighting System in accordance with Ariff MOD. DO-365-33-104 Revision 1: Introduction of new helicopter configurations.</td>
<td>- Initial revision Ariff Flight Manual Supplement: FMS-365-33-104-01. Rev. 14 Afr. AS 36983 NVIS Lighting System Installation: MOD. DO-365-33-104 Rev. 1 Ariff Instruction for Continued Airworthiness. ICA-365-33-104 Rev. 0 Ariff Master Document for List for DO - 365-33-104 Rev 0 Revision 1A.</td>
<td>Refer to applicable Flight Manual Supplement (FMS) for latest equipment mandatory for Night Vision Imaging System (NVIS) certification. Rotorcraft modified by this design change employs Night Vision Imaging System (NVIS) certified Rotorcraft commonly known as NVIS lighting, which has been certified to ensure the aircraft...</td>
<td>08/02/2024</td>
<td>Active</td>
<td>10020077</td>
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</table>
Installation of one twenty (22) gallon auxiliary fuel tank in each outboard wing panel in acc. with FAA STC SA1614WE.

The Certification Basis for the original product remains applicable. 

The requirements for environmental protection and the associated certified noise/ or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.

The Installation and conditions of FAA Aircraft Specification A-799, FAA Data Sheets 3A12, 3A17, or FAA Data Sheets AEU and A19EU or EASA TCDSIM.A.051 apply, except as outlined below of this STC. 

Installation of NVIS Lighting System in

Installation of NVIS Lighting System in accordance with AIRplus. MOD. DO-365-33-101.

The Certification Basis for the original product remains applicable to this certificate/ approval. 

The requirements for environmental protection and the associated certified noise/ or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.

Installation of one twelve (12) gallon associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.

Installation of a pylon mounted ejector panel in acc. with FAA STC ST01567LA.

The Certification Basis for the original product remains applicable to this certificate/ approval. 

The requirements for environmental protection and the associated certified noise/ or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.

Installation of two 12 gal. auxiliary tanks in wing storage areas based on known previouslyinstalled modification and any other interrelationships between this modification it must be determined that there is no adverse effect upon the aircraft handling, aerodynamic efficiency and/or repair will introduce no adverse effect on the airworthiness of the original product. 

Installation: MOD. NVIS Lighting System document approved 101-01, Rev. 0 or later revisions of the above listed documents approved by EASA.

Installation of New J Class Crew Rest for Continental Airworthiness.

Design Change Statement do. N° DC-0194 Issue 1 dated 20.05.2011 or later revisions of the above listed documents approved by EASA.

Installation of New J Class Crew Rest for Continental Airworthiness.

Installation of New J Class Crew Rest for Continental Airworthiness.

Installation of New J Class Crew Rest for Continental Airworthiness.
<table>
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<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
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<th>Models</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>10035124</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST03070AT</td>
<td>GIV-X</td>
<td>Installation of a Honeywell LSZ-860 Lightning Sensor System.</td>
<td>The Certification Basis for the original product remains applicable. Only previously approved modifications are incorporated. It is determined by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse effect.</td>
<td>10035125 Installation in accordance with Gulfstream Aerospace Corporation Index List GC41225800, Revision F, dated March 28, 2005. Operation in accordance to Gulfstream Flight Manual supplement ref GC4122SM00 dated 05 April, 2005, &quot;Honeywell LSZ-860 Lightning Sensor System&quot; or later revisions of the above list.</td>
<td>27/05/2011 Active</td>
<td>10035124</td>
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<td>10035125</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST03566AT-D</td>
<td>GIV-X</td>
<td>Installation of a Custom Cabin Interior and completion of Aircraft Serial Number 4121.</td>
<td>The Certification Basis for the original product remains applicable. Only previously approved modifications are incorporated. It is determined by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse effect.</td>
<td>27/05/2011 Active</td>
<td>10035125</td>
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<td>10035128</td>
<td>LEARJET inc./DB/ A BOMBARDIER AIRCRAFT SERVICES</td>
<td>FAA STC SA06526DE</td>
<td>BD-700-1A10</td>
<td>Modification of the Forward Water Fill Panel</td>
<td>EASA TCDS IM.A.009 The Certification Basis for the original product remains applicable. Only previously approved modifications are incorporated. It is determined by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse effect.</td>
<td>27/05/2011 Active</td>
<td>10035128</td>
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<td>10035154</td>
<td>BEECHCRAFT CORPORATION</td>
<td>FAA STC SA0526DI</td>
<td>390 (PREMER IA) 390 (PREMER I)</td>
<td>Installation of an AirCell telephone system.</td>
<td>The Certification Basis for the original product remains applicable. Only previously approved modifications are incorporated. It is determined by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse effect.</td>
<td>15/11/2012 Active</td>
<td>10035154</td>
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<td>10035155</td>
<td>STANDARD AERO</td>
<td>FAA STC ST08630CH-D</td>
<td>EMB-145LF, EMB-145XR EMB-145ER, EMB-145MR, EMB-145, EMB-145MP, EMB-145EP, EMB-135KE, EMB-135K, EMB-135ER, EMB-135LR</td>
<td>Installation of a Honeywell KHF-1050 HF System.</td>
<td>The Certification Basis for the original product remains applicable. Only previously approved modifications are incorporated. It is determined by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse effect.</td>
<td>None.</td>
<td>30/05/2011 Active</td>
<td>10035155</td>
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<td>BRISTOW HELICOPTERS LIMITED</td>
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<td>Qualification of the Garmin 400 Series GPS for enroute navigation and precision approach.</td>
<td>CS2973-the requirements for environmental protection and the associated certified noise and or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>30/05/2011</td>
<td>Active</td>
<td>10035156</td>
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<tr>
<td>AVIONIK STRAUSSING ENTWICKLUNGS GmbH</td>
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<td>Installation of dual GPS/NAV/COMM Garmin GNS430/530(A)W.</td>
<td>The Certification Basis for the original product and the following: additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23, Amendment 2 (not required by 21A.101) The requirements for environmental protection and the associated certified noise and or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>31/05/2011</td>
<td>Active</td>
<td>10035162</td>
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<td>STC TWENTY ONE LIMITED</td>
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<td>Introduction of Number 2 Galley Assembly</td>
<td>The Certification Basis for the original product and the following: additional or alternative airworthiness requirements are applicable to this certificate/approval FAR 25.1529 at Amendment 00 Instructions for Installation: ASR-2010-058-FMS-05-00, Amendment 00 Instructions for Continued Airworthiness: ASR-2010-058-IFR-10-00, Amendment 00 Instructions for Continued Airworthiness: ASR-2010-058-ICA-11-00, Amendment 00 or later revisions of the above.</td>
<td>None.</td>
<td>01/06/2011</td>
<td>Active</td>
<td>10035173</td>
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<td>SABENA TECHNICS BQC</td>
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<td>Aeroshell STC 0490-1 Airlift ST3100 IATCOM Installation on A300-600LR, This STC covers the installation of a Satellite Telephone System/Transceiver, a Telephone Adapter with Display, a Base cradle for the Cordless Handset and a Satellite antenna in accordance with Master Data List Ref.0490-01-A-2103 Rev...</td>
<td>The Certification Basis for the original product and the following: additional or alternative airworthiness requirements are applicable to this certificate/approval - EASA - Certification Specification CS 25 Amdt 4 - Special condition SIC 901 (Enhanced airworthiness programme for aeroplane systems - E1)...</td>
<td>None.</td>
<td>01/06/2011</td>
<td>Active</td>
<td>10035182</td>
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<td>STC TWENTY ONE LIMITED</td>
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<td>Introduction on EMB-145 of Saflite photoluminescent FPEEPMS</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise/emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>03/06/2011</td>
<td>Active</td>
<td>10035193</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>PRECISE FLIGHT, Inc.</td>
<td>10035215</td>
<td>REV. 1</td>
<td>FAA STC SAE262NM</td>
<td></td>
<td>Installation of Precise Flight Potulite Control System Rev. 1 updates the applicability to DHC-7 and DHC-8 models.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Engineering Design List ref. 250/0000 rev. A, dated 22 February 2010 - Instructions for Continued Airworthiness ref. 000PMAN/0002 rev. G, dated 06 May 2020 - Aircraft Flight Manual Supplement listed on FAA AML “N” SA2622NM dated 06 December 2004. Installation Manual listed on FAA AML “N” SA2622NM dated..</td>
<td>This approval is restricted to the following Potulite units: 2400-25-4-4-4 / 2400-25-4-4-1 / 2400-25-4-4-2 / 2401 / 2401. This approval does not cover the physical installation of the system. Physical installation of the system in the aircraft should be duly substantiated by an appropriate entity and app...</td>
<td>16/11/2021</td>
<td>Active</td>
<td>10035215</td>
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<tr>
<td>ASI INNOVATION</td>
<td>10035229</td>
<td>T206H</td>
<td>ASI-FAM-0002-01</td>
<td></td>
<td>Installation of Filter Kit on Cessna T206H.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Aircraft Flight Manual Supplement ASI-AFM-0002-03 dated 22 April 2011 ASI-SDL-0002-05 Maintenance Planning Supplement ASI-MPS-0002-01 dated 17 March 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>Limitation to Cessna T206H serial numbers 445 and 446</td>
<td>07/06/2011</td>
<td>Active</td>
<td>10035229</td>
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<tr>
<td>DANSK FLY ELEKTRONIK ApS</td>
<td>10035243</td>
<td></td>
<td>CESSNA 414, 414A</td>
<td></td>
<td>Installation of Garmin GTX330D Transponder</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Design Order Package DFE-DO-0439 including Aircraft Flight Manual Supplement (DFE-DO-0439-AMF) and Instructions for Continued Airworthiness (DFE-DO-0439-ICA) or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
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<td>10035243</td>
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<tr>
<td>KUERZI AVIATION AG</td>
<td>10035249</td>
<td>PC-6-B2-H4</td>
<td></td>
<td></td>
<td>Cold weather operation package (230°FAC or 110°FAC) by installation of: One TANS20200 Battery Heater system - One TANISTAS300 Turbine Preheat system - One Power receptacle in the cabin front right.</td>
<td>CAR 3, Amndt. 3-1 through 5.7The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable: The certificate/approval(s) or ESFs: For this change, voluntary compliance to CS23 Amndt. 0 - The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List: MDL-201110187, dated 08.06.2011 Project Installation Manual: PAM-201110187, dated 08.06.2011 Instruction for continued airworthiness: ICA-201110187, dated 08.06.2011 Airplane Flight Manual Supplement: FMS-201110187, dated 08.06.2011 or later revisions of the above listed documents.</td>
<td>S/N: all</td>
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<td>ALIDAUNA S.R.L.</td>
<td>10035253</td>
<td>A109A II</td>
<td>MB 022(A)/2010 &quot;Installation of Alternate Static Pressure Source for the Indicator: Altimeter, Airspeed and Vertical Speed&quot;</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement Doc. FMS/A01/002 Rev. 0 or later revisions of the above listed documents approved by EASA/BDT DocDO/ALLI-002 Rev. 0 Maintenance Manual Supplement Doc. MMS/A01/002 Rev. 0</td>
<td>None</td>
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<td>SPECTRUM AEROMED, LLC/</td>
<td>10035259</td>
<td></td>
<td>FAA STC SA20935CH</td>
<td>441</td>
<td>Air Ambulance Conversion in accordance with Spectrum Aeromed Drawinglist</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificate/approval(s) or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA approved Master Drawing List Doc. No. 66786D, Rev. - dated 16th February 2010 (or later) &gt; FAA approved Aircraft Flight Manual Supplement Doc. No. 8078FS, Rev. - dated 12 July 2010 or later revisions of the above listed documents approved by EASA. In accordance with EASA ED Decision 2004/04/CF (or subsequen...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce unacceptably adverse impact on the airworthiness of the product.</td>
<td>23/04/2019</td>
<td>Active</td>
<td>10035259</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>No.</th>
<th>STC Holder</th>
<th>STC Type</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<th>System Record</th>
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<tr>
<td>10035276</td>
<td>HELI-ONE</td>
<td>TCA STC- SH05-46</td>
<td>S-61N</td>
<td>Sealed Lead Acid Battery Installation</td>
<td>CAP-7 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Supplement CHS1-24-016 Revision A(Revises the above listed documents approved by EASA in accordance with EASA ED Decision 2004/020/C.F or subsequent revisions of this decision).</td>
<td>15/06/2011</td>
<td>Active</td>
<td>10035276</td>
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<tr>
<td>10035279</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST00785AT-D</td>
<td>GIV-SP</td>
<td>Aircraft Completion and Installation of an Airplane Interior</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation of an Airplane Interior in accordance with GulfstreamAerospace Corporation Index List GC520005280, Revision D, dated September 23, 2016.</td>
<td>10/06/2011</td>
<td>Active</td>
<td>10035279</td>
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<tr>
<td>10035287</td>
<td>SAFRAN ELECTRONICS &amp; DEFENSEBUSINESS SS UNIT - HELICOPTERS DATAMANAGEMENT NT SOLUTIONS &amp; SERVICES</td>
<td>REV. 9</td>
<td>EC 130 B4, EC 130 T2 AS 350 B2, AS 350 B3</td>
<td>Installation of Flight Data Monitoring System HELICOM V2 NUM or HELICOM V2+ NUM.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Rotorcraft Flight Manual Supplement N14.</td>
<td>27/02/2020</td>
<td>Active</td>
<td>10035287</td>
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<tr>
<td>10035295</td>
<td>AIRCRAFT DESIGN CERTIFICATION GmbH</td>
<td>REV. 2</td>
<td>L-13 “BLANÍK”</td>
<td>Modification and inspection program for compliance with EASA-2010-0165-E.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification and inspection program for compliance with EASA-2010-0165-E.</td>
<td>17/07/2018</td>
<td>Active</td>
<td>10035295</td>
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<td>10035296</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST01868A</td>
<td>B767-300</td>
<td>Installation of Rockwell Collins CMU-900 Communications Management System, APX-600 Airplane Personality Module, ICD-800 Integrated DataLink Controller &amp; Tidéalyn TP-4499 Printer.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this approval.</td>
<td>None.</td>
<td>14/06/2011</td>
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<td>STC Holder</td>
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<tr>
<td>Spectrum Aeromed, LLC/ SPECTRUM AEROMED</td>
<td>FAA STC SA20897CH</td>
<td>A320-214</td>
<td>Air Ambulance Conversion according FAA STC SA20897CH.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>CS 23</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/04/2019</td>
<td>Active</td>
<td>10035308</td>
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<tr>
<td>ACS-NAI Ltd.</td>
<td>TCCA STC C-LSA11-075/D</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>TCCA STC No. C-LSA11-075/D Issue 1 dated 13 May 2011, ACS-NAI Ltd. Modification Data Summary G789000 Revision 0 or later revision and/ or revision 0 or later revision of the above listed documents approved by EASA in accordance with EASA.</td>
<td>Restricted to ASN 9403. This system is intended to provide internet connectivity and email services to the airplane’s cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require an examination of the certification basis. Embodiment of EASA STC No. EASA.IM-SA2971SO.</td>
<td>15/06/2011</td>
<td>Active</td>
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<tr>
<td>Northern Avionics S.R.L.</td>
<td>A01095P</td>
<td></td>
<td>Installation of a Relay box and Control Panel for ICS VIP application.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Rotorcraft Flight Manual Supplement for ICS VIP Miscellaneous Control Panel (KIT P/N KNAV-418), RIMRS NAV 438 Rev. 0 or later revision of the above listed documents approved by EASA.</td>
<td>Required pre-modification: Agusta electrical/mechanical provision n°100-B910-30.</td>
<td>05/08/2011</td>
<td>Active</td>
<td>10035325</td>
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<tr>
<td>PEMCO World Air Services</td>
<td>FAA STC SA2971SO</td>
<td>B737-300</td>
<td>Installation of a Cargo Loading System in accordance with MDL 2337Rev X dated 12/07/2010. Modification of the STC FAA SA2971SO to increase from 121 pax to 148 pax according to drawing 90485-5075607.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Project Specific Certification Plan for update of STC SA2971SO to include 148 passenger configuration of the Quick Change Handling/ Load System for B737-300 S/N 23522 and 23523. - Report No. ER-096 rev 3 dated 09/09/2011 (Drawing 90485-5075607 as described in master Drawing List B737-200 and B737-200A).</td>
<td>None</td>
<td>16/06/2011</td>
<td>Active</td>
<td>10035332</td>
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<tr>
<td>DOA21 Limited</td>
<td>A320-214</td>
<td></td>
<td>Custom Interior A320 JubileeRev. 1 Transfer of STC holder role to DOA21 Ltd.</td>
<td>The Type and OBS Certification Bases for the original product are amended by the following additional or alternative airworthiness requirements: The following EASA Certification Review Items (CRIs) are part of the certification basis for this design change: - Deviation CRI D-01 Issue 1 &quot;Installation of...</td>
<td>STC Twenty One Limited documents - Master Document List S21.MX-0509 initial issue dated 02nd June 2011, - Aircraft Flight Manual Supplement (AFM) Supplement S21.FMS-0079 Issue 1, - For documents and supplement manuals related with instructions for continued airworthiness see references on Master Document ListS2...</td>
<td>This design change approval is applicable on Aircraft model Airbus A320-214, aircraft serial number 4198. This STC has associated a supplement to the original TC Holder’s EWIS instructions for Continued Airworthiness. This supplement is included in the document JUBL-CRR-400-EWA. *Instructions for continu...</td>
<td>24/10/2023</td>
<td>Active</td>
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## List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<tr>
<td>10035406</td>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td></td>
<td>FALCON 2000</td>
<td>Indoor antenna installation. Structural justification of the installation of an Aircraft low profiled patch antenna on top of the fuselage between frames 17 and 18.</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List Doc. No. BOS/SR001-002 Rev. A dated 18.06.2011 Certification Compliance Sheet Doc. No. BOS/SR001-102-02 Rev A dated 26.01.2011.</td>
<td>Restricted to SN 1737. This approval only covers the structural installation of the antenna, installation and activation of any related Satcom system must be qualified and covered by another suitable approval.</td>
<td>21/06/2011</td>
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<td>10035406</td>
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<td>10035421</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td></td>
<td>FAA STC NO. ST04238AT-D</td>
<td>GV-SP (G550)</td>
<td>Installation of Cabin Interior and Aircraft Completion for Gulfstream GV-SP model Aircraft</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC52008724 Rev G, dated 06 May, 2011. Operation in accordance with Gulfstream Aerospace Corporation Airworthiness Manual Supplement No. GC52000M149 Rev -. dated 06 June, 2011.</td>
<td>None.</td>
<td>21/06/2011</td>
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<td>10035421</td>
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<td>10035429</td>
<td>DART AEROSPACE Ltd.</td>
<td>REV. 2</td>
<td>TCCA STC SH11-1</td>
<td>407 206L, 206L-1, 206L-3, 206L-4</td>
<td>D407-797 Heli Utility Basket and Helil Access Step Installation, Revised STC 10035429 to add D407-797-311 Heli AccessStep.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable. The applicant complied voluntarily with the following paragraphs of FAR27 amendment 27.49, 27.51, 27.57, 27.29, 27.49, 27.51, 27.29, 27.49, 27.51, ...</td>
<td>FMS-D407-797 revision A dated 12 January 2011 - Flight Manual Supplement later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>28/11/2011</td>
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<td>10035429</td>
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<tr>
<td>10035437</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A319-133</td>
<td>Satcom and LAN / WLAN System Installation</td>
<td>A319-133</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable. The applicant complied voluntarily with the following paragraphs of FAR27 amendment 27.49, 27.51, 27.57, 27.29, 27.49, 27.51, ...</td>
<td>- Master Data List, ref. doc. MDL-BL01-2300, Rev. A, dated 1st June 2011. Certification Plan, ref. doc. CP-BL01-2300, Rev. B, dated 20th June 2011. Compliance Check List, ref. doc. CCL-BL01-2300, Rev. B, dated 23rd March 2011. Declaration of Compliance - Project Satcom and LAN / WLAN System Installation...</td>
<td>Not Applicable.</td>
<td>22/06/2011</td>
<td>Active</td>
<td>10035437</td>
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<tr>
<td>10035456</td>
<td>AUWINK STRAUBING ENTWICKLUNGS GmbH</td>
<td>F406</td>
<td>Cockpit Upgrade basing on dual Garmin G600 and GNS430W</td>
<td>F406</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable. The applicant complied voluntarily with the following paragraphs of FAR27 amendment 27.49, 27.51, ...</td>
<td>- Master Data List, ref. doc. MDL-BL01-2300, Rev. A, dated 1st June 2011. Certification Plan, ref. doc. CP-BL01-2300, Rev. B, dated 20th June 2011. Compliance Check List, ref. doc. CCL-BL01-2300, Rev. B, dated 23rd March 2011. Declaration of Compliance - Project Satcom and LAN / WLAN System Installation...</td>
<td>Only applicable for Aircraft SN 097 and up.</td>
<td>24/06/2011</td>
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<td>10035456</td>
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<td>STC Holder</td>
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<td>Model(s)</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
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<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>STC N°ST105835C-D</td>
<td>2 F900EX 1 F2000EX</td>
<td>Gatley pocket door installation/validation of FAA STC N°ST105835C-D</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>DF J Certification Data List doc. N° F90-01005 Revision AR dated April 03, 2011. Instructions for Continued Airworthiness doc. N° ICA- F90-01005-500 Revision Original, dated January 25, 2011. Airplane Flight Manual Supplement doc. N° 7387-1 dated July 18, 2007, or later revisions of the above listed document...</td>
<td>Approval is limited to the “-501” configuration.</td>
<td>24/06/2011</td>
<td>Active</td>
<td>10035461</td>
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<tr>
<td>BRISTOW HELICOPTERS LIMITED (EASA.R.01487)</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBLITY</td>
<td>S-92A</td>
<td>Installation of Rockwell Collins TCAS II system. Rev. 1 covers modification of Rockwell Collins TCAS-4000 TCAS II update to version 7.1 software. Modification S92/03.5702</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>18/07/2012</td>
<td>Active</td>
<td>10035471</td>
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<td>SKYE AVIONICS Ltd.</td>
<td></td>
<td></td>
<td>BELL 407</td>
<td>Installation of “Check Engine Instruments” Audio Alert System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>AER-FMS-023 revision 01 dated 29 April 2011 - Flight Manual Supplement - later revisions of the above listed documents approved by EASA in accordance with EASA - Flight Manual Supplement No 4.</td>
<td>None</td>
<td>27/06/2011</td>
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<td>KUERZI AVIATION AG</td>
<td></td>
<td></td>
<td>DHC-6-300</td>
<td>Installation of two Aspen Electronic Flight Display (ETF1000 Pro, one Garmin GNS530W (GPS, COM, NAV), one Garmin GNS430W(GPS, COM, NAV), one Garmin GTX330 (XPDR) and one Garmin GT580 (TAS)</td>
<td>The Certification Basis for the original product and the following modifications are applicable to this certificate/approval: F/A-23, Amtd. 02, F/A-23.1232 Amtd. 01.</td>
<td>Master document list MDL-201034120, Initial issue/Principal Installation Manual PM-201034120, Initial Issue/Flight Manual Supplement FMS-201034120, Initial issue/Instruction of Continued Airworthiness ICA-201034120, Initial issue for later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>27/06/2011</td>
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<td>NORTHERN AVIONICS S.R.L.</td>
<td></td>
<td></td>
<td>AB206B</td>
<td>NAV 427 “Installation of Alternate Static System”. This kit consists of the installation of an additional line to the original static line (if a 3-way valve/switch (with the air inlet port) to pick up the static pressure in the cabin when the external ports are blocked (for ice or snow ice). This is...</td>
<td>The Certification Basis for the original product and the following modifications are applicable to this certificate/approval: CS-27 Amtd. 2 (Ref. NAV-427 Rev. 0 Sec. VI). The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>RFMS NAV 427 Rev. 0 (RFM Supplement &amp; Supplement al Manuale di Volo)/Technical Modification Bulletin NAV-427 Rev.</td>
<td>This STC is applicable only to the helicopters S/Ns from 8590 to 8758.</td>
<td>27/06/2011</td>
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<td>AVIONICS INTEGRATION &amp; ENGINEERINGCOP RORATION AG (AIEC)</td>
<td></td>
<td></td>
<td>EC135 P2+</td>
<td>GNS-530 &amp; GNS-430A Installation Out: Garmin GNS-530W &amp; GNS-430W. Garmin GNS-530 &amp; GNS-430A Configuration retrofit done by American Eurocopter under FAA Form 337 dated 11.12.2009 and FAA Form 8110-3 dated 11.12.2009</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List Doc. No.: EVS126-049 Rev. 0 (Instructions for Continued Airworthiness Doc. No. EVS126-049 Rev. 0 (Rotorcraft Flight Manual Supplement Doc. No. EVS126-071.01 Rev. 0)) or later revisions of the above listed documents approved by EASA</td>
<td>This EASA STC is only valid after EASA Validation of FAA STC SR910137RC-D (Extended R/H Side Instrument Panel) S/N 0964.</td>
<td>29/06/2011</td>
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### List of EASA Supplemental Type Certificates

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<td>MICHELIN NORTH AMERICA, INC.</td>
<td>10035563</td>
<td>ST05942AT</td>
<td>05/07/2011</td>
<td>10035563</td>
<td>CL-600-2D4 CL-600-2C10, CL-600-2D15</td>
<td>Installation of Michelin's tires P/N 029-901-0 on the main gear and/or P/N 036-869-1 on the nose landing gear of Bombardier CL-600-2C10 CL-600-2D15 CL-600-2D24 aircraft wheel assembly.</td>
<td>EASA.IM.A.023 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated listed noise and/or emissions levels of the original product are unchanged and remain applicable to the certificate/approval.</td>
<td>FAA STC No. SA03642AT dated 10 March 2016 Aircraft Braiding Systems Corporation (AABC) Component Maintenance Manual(OMM) AP-83, Revision 1, dated 20 April 2004, AP-734, Revision 1 dated 1 April 2003, AP-756 dated 1 February 2001, and Michelle North America Inc. Service Bulletin STC SB 2008-10-15 Rev...</td>
<td>30/06/2011</td>
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<td>DASSAULT FALCON SERVICE</td>
<td>10035568</td>
<td>REV. 2</td>
<td>03/07/2024</td>
<td>10035568</td>
<td>MF900, F900EX</td>
<td>WiFi activation on MF900 and F900EX. This revision 2 of the EASA STC 10035568 approves the extension of applicability of the change to the F900EX model, EASy version. NEW-PAGE</td>
<td>EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated listed noise and/or emissions levels of the original product are unchanged and remain applicable to the certificate/approval.</td>
<td>Master Data List DFS-DE-17-F9-9609 Issue D, dated 9 July 2015</td>
<td>13/07/2015</td>
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<td>BOMBARDIER INC.</td>
<td>10035585</td>
<td>REV. 1</td>
<td>2004/06/07</td>
<td>10035585</td>
<td>TCCA STC NO. C-LSA02-261/D</td>
<td>Approval of Side Facing Divan Seating Limitations</td>
<td>EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated listed noise and/or emissions levels of the original product are unchanged and remain applicable to the certificate/approval.</td>
<td>Transport Canada STC No. C-LSA02-261/D Issue 2 dated 12 November 2003</td>
<td>06/07/2011</td>
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<td>AMES - AEROSPACE AND MECHANICAL ENGINEERING SERVICES INC. WALTERSTARZA GERSH</td>
<td>10035593</td>
<td>REV. 1</td>
<td>09/11/2012</td>
<td>10035593</td>
<td>BD-700-1A10</td>
<td>Approval of Side Facing Divan Seating Limitations</td>
<td>EASA.IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated listed noise and/or emissions levels of the original product are unchanged and remain applicable to the certificate/approval.</td>
<td>Transport Canada STC No. C-LSA02-261/D Issue 2 dated 12 November 2003</td>
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<td>AIRBUS HELICOPTERS, Inc.</td>
<td>10035617</td>
<td>FAA STC SH9171SW-D</td>
<td>none</td>
<td>10035617</td>
<td>SA-365 N1 AS 365 N2</td>
<td>Installation of Bendix/King EFIS 40 Electronic Flight Instrumentation System in accordance with AEC Drawing List 365A73-9055 Revision B dated 10/11/94 or Bendix/King EFIS 50 Electronic Flight Instrumentation System in accordance with AEC Drawing List 365A73-9056 Revision B dated 02/04/94</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated listed noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. SA03642AT dated 10 March 2016 Aircraft Braiding Systems Corporation (AABC) Component Maintenance Manual(OMM) AP-83, Revision 1, dated 20 April 2004, AP-734, Revision 1 dated 1 April 2003, AP-756 dated 1 February 2001, and Michelle North America Inc. Service Bulletin STC SB 2008-10-15 Rev...</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<td>STC TWENTY ONE LIMITED</td>
<td>F4</td>
<td>EA-0174</td>
<td></td>
<td>Customized Interior Completion - ASN 9308</td>
<td>STA-02395:11-36</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The requirements for environmental protection and the associated/certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>STC Twenty One Ltd Master Document No S21-MDL-0794 Issue 1STC Twenty One Ltd Flight Manual Supplement No. S21-FMS-004 Issue 1</td>
<td>Restricted to ASN 9308</td>
<td>05/07/2011</td>
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<td>HELI AUSTRIA GmbH</td>
<td>REV. 2</td>
<td>AS 355 F1, AS 355 F2</td>
<td></td>
<td>Compliance of C20R installation with BCAR Section G and confirmation of C20R to fly one engine inoperative profiles. Reason for Rev. 1: transfer of STC to Hel Austria GmbH/Reason for Rev. 2: limitation to helicopter serial numbers.</td>
<td>STC 027:11-36</td>
<td>The Type and OSD Certification Basis for the original product amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: BCAR Section G: The requirements for environmental protection and the associated/certificate noise and/or emissions...</td>
<td>STC 027:11-36</td>
<td>Limited to serial number 5161 &amp; S4738</td>
<td>12/04/2023</td>
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<td>BRUCE AEROSPACE Inc.</td>
<td>FAA STC</td>
<td>SA0011NM</td>
<td>CL-600-2B16</td>
<td>Installation of Floor Proximity Emergency Escape Path Marking System</td>
<td>EASA STC No. 0239:2011-12</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The requirements for environmental protection and the associated/certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Bruce Aerospace Master Drawing List CL600, Revision CE, Item 41, dated 12 December 2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>This system has been determined to comply with the applicable airworthiness requirements with the following operative light assemblies: - Floor Mounted Proximity Lighting System Exit identifiers and/or flood lights - 2 bulbs operative, light tracked modules - 90% operative (1 bulb in each 2 bulb group)</td>
<td>06/07/2011</td>
<td>Active</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>FAA STC</td>
<td>SA0011NM</td>
<td>A318-112</td>
<td>A318-112 - ELITE 16 MEN 4500 installation of a VIP Cabin Interior Completion.</td>
<td>EASA STC No. 0239:2011-12</td>
<td>Certification Basis in accordance with EASA Type Certificate Data Sheet65.04: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval: CRI A-5301-010 “EASA Supplemental Type Certification Basis...</td>
<td>Supplemental Type Definition Document, ref. Doc No STD000-1024M, Issue 01 revision 02, dated Jul 01/2011 – Airplane Flight Manual Supplement, ref. Doc No. A-00-10258-4FM, Issue 01 revision 02, dated Dec 31/2011 – Instructions for Continued Airworthiness, ref. Doc. No A-00-10258-4MA, Issue 01...</td>
<td>Not applicable</td>
<td>06/07/2011</td>
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<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>ST01239LA-D</td>
<td>CL-600-2B16</td>
<td>FAA STC ST01239LA-D, Complete Custom Aircraft Interior</td>
<td>EASA STC No. 0239:2011-12</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The requirements for environmental protection and the associated/certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>FAA STC No. ST01239LA-D, dated 02 December 2002, or later revisions of this STC, the installation of Bruce Aerospace EASA STC No. 1003638 and FAA STC No. SA0011NM (ref. EASA P/H010011S393) Floor Proximity Emergency Escape Path Marking System.</td>
<td>Restricted to ABN 541</td>
<td>06/07/2011</td>
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<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>FAA STC</td>
<td>ST01239LA-D</td>
<td>B200</td>
<td>Installation of a ADSB on a Beech 200</td>
<td>FAA STC ST01239LA-D, dated 02 December 2002</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The requirements for environmental protection and the associated/certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>FAA STC ST01239LA-D, dated 02 December 2002</td>
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<td>10035662</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>Installation of a ALT4000 on a Beech 200</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification noted/ward or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>none</td>
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<td>10035666</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>Installation of a DME4000 on a Beech 200</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification noted/ward or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>none</td>
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<td>10035671</td>
<td>ADAC LUFTFAHRT TECHNIK GMBH</td>
<td>Installation of fixed Landing Lights on the forward cross tubes in the MDL. EC135 C1-0172-001-LMD and on based on ALT change project EC135-0172.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement, EC135-0107-2-025-AM5-00 Rev. 0 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>08/07/2011</td>
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<td>10035677</td>
<td>MARSHALL OF CAMBRIDGE AEROSPACELIMITED</td>
<td>Installation of an Auxiliary Fuel Centre Tank system on A318-112 aircraft in combination to the aircraft.</td>
<td>Certificate of Design doc. N° CD/A318/2179/14/0255 Issue 1 dated 22/05/2015</td>
<td>None</td>
<td>31/06/2016</td>
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<td>10035678</td>
<td>CALEDONIAN AIRBORNE SYSTEMS LTD</td>
<td>Installation of Role Equipment for Maritime Patrol Training</td>
<td>Restricted to A318-112 aircraft with Airbus modifications 151462 and 151387</td>
<td>None</td>
<td>11/07/2011</td>
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<td>10035689</td>
<td>TENENCIA Ltd.</td>
<td>CAS UK RESPONSIBIL ITY</td>
<td>BAE 146 SERIES 200</td>
<td>EWS Assessment for VIP Cabin Interior</td>
<td>The Certification Basis for the original product and the following: additional or alternative airworthiness requirements are applicable to this STC/certificate/approval. Electronic Wiring Interconnection System (EWIS) requirements in CS-25 Amendment 8 Appendix H, paragraph HC25.5 and AMC Appendix H paragraph...</td>
<td>Modification TEN 664 EWS Assessment for VIP Cabin Interior EASA STC 10027954. Tenencia Ltd document number TEN 664, Issue 1 dated 29 January 2011. Aviation Traders Limited VIP Cabin Interior Modification Summary Sheet, basis 2 for MOD-111953-0001 or later revisions of the above listed documents...</td>
<td>12/07/2011</td>
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<td>10035701</td>
<td>BOMBARDIER INC.</td>
<td>REV. 1</td>
<td>TC-CA STC NO. SA11-30</td>
<td>Installation, Portable Water System UV</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/approval remain applicable to this certificate/approval...</td>
<td>Transport Canada STC No. SA11-30 Issue 1 dated 13 April 2011. ACS-NAI Ltd. Modification Data Summary GS0000 Revision NC or later approved revision ACS-NAI Ltd.</td>
<td>Prior to the installation of this modification it must be determined that the aircraft interior has been approved by the issuance of an EASA STC that corresponds to one of the Transport Canada STCs listed on Transport Canada STC No. SA11-30.</td>
<td>14/07/2011</td>
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<td>10035704</td>
<td>BOMBARDIER INC.</td>
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<td>TC-CA STC NO. C-LSA11-025/D</td>
<td>Installation, Avionics Systems</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/approval remain applicable to this certificate/approval...</td>
<td>Transport Canada STC No. C-LSA11-025/D Issue 1 dated 07 July 2011. ACS-NAI Ltd. Modification Data Summary G720000 Revision NC or later approved revision ACS-NAI Ltd.</td>
<td>1. Restricted to ASNs 9395, 9404, 9411, 9417. 2. Embodiment of EASA STC No. 10034902 (Transport Canada STC No. C-LSA11-024/D)</td>
<td>13/07/2011</td>
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<td>10035711</td>
<td>BOMBARDIER INC.</td>
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<td>TC-CA STC NO. C-LSA11-022/D</td>
<td>Installation, Complete Custom Aircraft Interior - ASNs 9395, 9404, 9411, 9417</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/approval remain applicable to this certificate/approval...</td>
<td>Transport Canada STC No. C-LSA11-022/D Issue 1 dated 13 May 2011. ACS-NAI Ltd. Modification Data Summary G720000 Revision NC or later approved revision ACS-NAI Ltd.</td>
<td>1. Restricted to ASNs 9395, 9404, 9411, 9417. 2. Embodiment of EASA STC No. EASA IM.A.02172</td>
<td>13/07/2011</td>
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<td>10035712</td>
<td>BOMBARDIER INC.</td>
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<td>TC-CA STC NO. C-LSA11-023/D</td>
<td>Installation, Medical Evacuation Configuration Provisions</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/approval remain applicable to this certificate/approval...</td>
<td>Transport Canada STC No. C-LSA11-023/D Issue 1 dated 13 May 2011. ACS-NAI Ltd. Modification Data Summary G720000 Revision NC or later approved revision ACS-NAI Ltd.</td>
<td>1. Restricted to ASNs 9395, 9404, 9411, 9417. 2. Embodiment of EASA STC No. 10035711 (refers to PN 0010007349 (Transport Canada STC No. C-LSA11-022/D))</td>
<td>13/07/2011</td>
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<td>MARSHALL OF CAMBRIDGE, AEROSPACE &amp; TERRORISM</td>
<td>REV. 2</td>
<td>FAA STC ST100069C-D</td>
<td>A340-313</td>
<td>Iberia Airways Interior Reconfiguration</td>
<td>The Certification Basis for the original product and the following:</td>
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<td>ASSOCIATED AIR CENTER, L.P.</td>
<td>REV. 1</td>
<td>FAA STC ST100066C-D</td>
<td>A319-115</td>
<td>Installation of an Executive Style Interior/ESA validation of FAA STC ST100069C-D</td>
<td>The Certification Basis for the original product and the following:</td>
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<td>NORTHERN AVIONICS S.R.L.</td>
<td>REV. 1</td>
<td>AW100SP</td>
<td>BD-700-1A11</td>
<td>Installation of In-flight Entertainment System (IES) (revised by kIPN KNAV-419 and KNAV-420/Certificate rev. 1 issued due to change of Modification Bulletin/Revision level)</td>
<td>The Certification Basis for the original product remains applicable to this certificate.</td>
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<td>BOMBARDIER INC.</td>
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<td>TCCA STC NO. 5A05-60</td>
<td>PA-42-720</td>
<td>Installation, Complete Custom Aircraft Interior: ASN 9418</td>
<td>EASA TCDS No IM.A-009/15-001 - The Certification Basis for the original product remains applicable to this certificate.</td>
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<td>AIR ALLIANCE GmbH</td>
<td></td>
<td>BM-42-314</td>
<td>PA-42-720</td>
<td>Extension of pressurized fuselage life limit to account for the change in operational usage from Altai Airlines Training to Normal Usage by owner.</td>
<td>The Certification Basis for the original product remains applicable to this certificate.</td>
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<td>BOGGI S.R.L.</td>
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<td>AS365N3</td>
<td>PA-42-720</td>
<td>Installation of Draeger Ti 500 &quot;Glue-Stopper&quot; incubator on AS 365N3 (ref. Change 365-M003-00).</td>
<td>The Certification Basis for the original product remains applicable to this certificate.</td>
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<td>STC Holder</td>
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<td>BRISTOW U.S. L.L.C.</td>
<td>REV. 1</td>
<td>FAA STD SR06379C</td>
<td>S-76A, S-76C</td>
<td>Installation of an Automatic Deployable Emergency Transmitter (ADELT) (Rev. 1): Changes 'System interface Unit' p/n to 503-24-A to comply with EASA AD 2014-00959R1.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
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<td>29/06/2017</td>
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<td>EMTEQ HOLDING CORP</td>
<td>REV. 1</td>
<td>FAA STD ST03064AT</td>
<td>CL-600-2B19</td>
<td>Installation of EMTEQ LED Anti-collision/Navigation Light in the Bombardier Regional Jet Series 100/300/400 Aircraft.</td>
<td>The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product have not changed and remain applicable to this certificate/approval.</td>
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<td>02/03/2016</td>
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<td>INNOTECH COMPANY INC</td>
<td>REV. 1</td>
<td>FAA STD ST03064AT</td>
<td>CL-600-2B19</td>
<td>Installation of EMTEQ LED Anti-collision/Navigation Light in the Bombardier Regional Jet Series 100/300/400 Aircraft.</td>
<td>The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product have not changed and remain applicable to this certificate/approval.</td>
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<tr>
<td>SABENA TECHNICS BOC</td>
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<td>FAA STD B77-300, B777-300ER</td>
<td>B777-300, B777-300LR</td>
<td>0555-01: Provisions of OnAir GSM/Internet System on B777.</td>
<td>The Certification Basis for the original product and the above additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
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<td>25/07/2011</td>
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<td>FIELD AVIATION COMPANY Inc.</td>
<td>REV. 1</td>
<td>FAA STD ST03064AT</td>
<td>CL-600-2B19</td>
<td>Installation of EMTEQ LED Anti-collision/Navigation Light in the Bombardier Regional Jet Series 100/300/400 Aircraft.</td>
<td>The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product have not changed and remain applicable to this certificate/approval.</td>
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## List of EASA Supplemental Type Certificates

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<th>STC Holders</th>
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<tr>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>FAA STC ST00522AT</td>
<td>B747-400 SERIES</td>
<td>Installation of a Collins Satcom update from 900 SDU to 906SDU (datalony) in accordance with Page Araf Master Drawing.1 fol 94-17, Initial Release, dated June 17, 1994, or later FAA approved Revision.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Certification Unit Page Araf 94-17. Initial Release, dated June 17, 1994 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/004/CF (or subsequent revisions of this decision).</td>
<td>none.</td>
<td>26/07/2011</td>
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<td>10035902</td>
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<tr>
<td>H4 AEROSPACE (UK) Ltd</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>777-200</td>
<td>Introduction of an Aerospace Cockpit Door and Cabin Surveillance on Boeing 777-200.</td>
<td>The Certification Basis for the original product and the following addtional or alternative airworthiness requirements are applicable to this certificate/approval. CRJ-1-ICA on EWIST. The certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC ST10903SC-D</td>
<td>none</td>
<td>27/07/2011</td>
<td>Active</td>
<td>10035914</td>
<td></td>
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<tr>
<td>STANDARD AERO BUSINESS AVIATIONSERVIC ES, L.L.C</td>
<td>FAA STC ST10903SC-D</td>
<td>A319-133</td>
<td>Modification of an Executive Interior.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for the environmental protection and the associated certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Document List, Associated Air Center document No. 2708000000, Revision A dated 30 November 2017. 2. Airplane Flight Manual Supplement (AFMS) document No. 2708020100, Revision IR dated 27 October 2010. 3. Instructions for Continued Airworthiness, Associated Air Center document No. 2708030000. (COTS) Nonessential Equipment, document number 1668/2002, Revision IR dated 13 October 2010, are considered inadequate for duplication and limited to aircraft Serial Number 910 only. The wireless LAN installation is intended to p...</td>
<td>The items identified in document &quot;Summary of Commercial Off The Shelf (COTS) Nonessential Equipment&quot;, document number 1668/2002, Revision IR dated 13 October 2010, are considered inadequate for duplication and limited to aircraft Serial Number 910 only. The wireless LAN installation is intended to p...</td>
<td>13/02/2018</td>
<td>Active</td>
<td>10035915</td>
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<tr>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES,A.R.L.</td>
<td>FAA STC ST10903SC-D</td>
<td>A319-133</td>
<td>Modification of an Executive Interior.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The Certification Basis for the original product and the following addtional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>1. Master Document List, Associated Air Center document No. 2708000000, Revision A dated 30 November 2017. 2. Airplane Flight Manual Supplement (AFMS) document No. 2708020100, Revision IR dated 27 October 2010. 3. Instructions for Continued Airworthiness, Associated Air Center document No. 2708030000. (COTS) Nonessential Equipment, document number 1668/2002, Revision IR dated 13 October 2010, are considered inadequate for duplication and limited to aircraft Serial Number 910 only. The wireless LAN installation is intended to p...</td>
<td>The items identified in document &quot;Summary of Commercial Off The Shelf (COTS) Nonessential Equipment&quot;, document number 1668/2002, Revision IR dated 13 October 2010, are considered inadequate for duplication and limited to aircraft Serial Number 910 only. The wireless LAN installation is intended to p...</td>
<td>27/07/2011</td>
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<td>LIBERTY AEROSPACE INC.</td>
<td>FAA STC SA0348AT</td>
<td>XL-2</td>
<td>S-TEC system 30 autopilot installation on Liberty XL-2 series in accordance with the following: Option 1 for use with S-Tec Model ST-601 GPS Steering Converter Unit/Chippewa Aerospace Master Data List (MDL) n° XL-2210-01-0001, Revision A, dated November 20, 2008 or later FAA approved revision, or Options...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC ST10903SC-D</td>
<td>none</td>
<td>28/07/2011</td>
<td>Active</td>
<td>10035929</td>
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**Note:** The list above includes key details of EASA Supplemental Type Certificates as of the date of publication, but may not be exhaustive. For the most current information, please refer to the official EASA website or relevant databases.
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<tr>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>DHC-8-102/103/106</td>
<td>Installation of Right Hand Ground Air Connection/TCCA STC C-LSA11-043D, Issue 1</td>
<td>FAR 29 as per EASA M.191, Issue 4 (inc. SIC.01 ICA on EWS) The approval holder shall fulfill the obligations of Part 21, Paragraph 21.A109. The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification.</td>
<td>Installation of the Right Hand Ground Air Connection must be in accordance with Transport Canada Civil Aviation (TC CA) approved Field Aviation Modification Instruction 128120, Revision IR, dated July 21, 2010. Maintenance of the Ground Air System is unchanged from the instructions provided in the Bombardier A330 MRTT FSTA Modification</td>
<td>OEM installed APR must be removed in accordance with EASA STC 10035876 and TCCA STC SA98-30, Issue 3.</td>
<td>29/07/2011</td>
<td>Active</td>
<td>10200030</td>
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<tr>
<td>CESSNA DUESSELDORF CITATION SERVICE CENTER GmbH</td>
<td>BD-700-1A11</td>
<td>Cabin Reconfiguration</td>
<td>EASA M.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification. Noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Jat Aviation Declaration of Compliance No. 2011-3 dated 27/07/11, Jat Aviation Engineering Order No. 2011-3 Revision A or later revisions of the above listed documents approved by EASA.</td>
<td>Restricted to ASN 9168A prerequisite for this approval is Transport Canada STC No. SA06-60/FAA STC No ST0240NY as applicable to ASN 9168. However EASA has not approved this STC for ASN9168 and if this aircraft is to be entered on EU register then the STC must be approved by EASA For Non-EU Country...</td>
<td>29/07/2011</td>
<td>Active</td>
<td>10200041</td>
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<tr>
<td>AIRBUS DEFENCE AND SPACE S.A. (EASA.A.S.03607)</td>
<td>A330-243</td>
<td>A330 MRTT FSTA Modification STC/Rev. 1 introduces the following changes:- Deletion of previous limitation no. 4 on Autoland / CAT III function. Replacement of temporary conservative noise levels by final values- Partial deletion of previous limitation no. 1 on occupation of seats in the aeromedical area...</td>
<td>The Type and OOD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS 25.0853 Initial/Issue for affected areas by change DCV/1086-E This certificate/approval involves a change...</td>
<td>- Conversion Definition A330 MRTT FSTA (STC2) document ref. ME-FU- DT110148 issue D-ALS Part 1 Supplement “A330 FSTA Safe-Life Airworthiness Limitations Supplement for EASA STC2”, doc. no. ME-FU-DT-110099 issue F, “TE-FU-DT-22004 “Variation 01 to Issue F of A330 FSTA Safe-LifeAirworthiness Limitations Supplement”</td>
<td>* - Based on the conditions established in CRI G-01 with respect to the definition of the Airworthiness Limitation Section of the ContinuousAirworthiness the following limitations apply. As the modifications of this STC are intended for military role, incorporation in the civil register of the aircraft...</td>
<td>17/01/2024</td>
<td>Active</td>
<td>10200045</td>
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<tr>
<td>SPECTRUM AEROMED, LLC/ (aka SPECTRUM AEROMED)</td>
<td>FAA STC SA0930CH</td>
<td>CESSNA 206, 208B Air Ambulance Conversion in accordance with Spectrum Aeromed Drawing list</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise/ and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>FAA approved Master Drawing List, Doc. No. 6006DL, Rev. 3, dated 24 Jan 2011 &lt;(&gt;) FAA approved Aircraft Flight Manual Supplement, Doc. No. 6006FMS, Rev., dated 30 Nov 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a positive effect upon the airworthiness of the product.</td>
<td>24/04/2019</td>
<td>Active</td>
<td>10200062</td>
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<tr>
<td>AIR ALLIANCE GmbH</td>
<td>LEARJET 35A</td>
<td>Installation of Honeywell GNS-XLS second FSM system.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification. Noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Air Alliance Master Document List 2011 006_L35_AA Form 2 Rev. 0 Aeronaut Flight Manual Supplement No: AA 2011_006_L35_0506 dated May 13, 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>Limited to 35A-421 and 35A-616.</td>
<td>03/08/2011</td>
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<td>EASA STC No.</td>
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<td>10035996</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>F4A STC: SR0506DCO</td>
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<td></td>
<td>MBB-BK 117-C2</td>
<td>Installation of tailboom mounted alt ballast in accordance with American/Eurocopter Drawing List 1454A2-8006.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Flight Manual Supplement 117FMS-22A, Rev. A dated 17 Mar 2006 later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
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<tr>
<td>10036025</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>PA 30</td>
<td></td>
<td>PFD ASPEN EFD 1000 PRO INSTALLATION on Piper PA 30.</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Date List + INSTALLATION PFD ASPEN EFD 1000 PRO SUR PIPER PA 30.vRef. AE 1010 MDL Rev. 5</td>
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<td>10036039</td>
<td>MID-CANADA MOD CENTER</td>
<td>CESSNA 550, S550</td>
<td></td>
<td>Installation of Rockwell Collins Proline 21 Integrated Display System (new VHF communication transceivers, Navigation Receivers with integrated ADS, Distance Measuring Equipment Transceivers and Mode S Transponders with Elementary Surveillance capability). Rockwell CollinsTADS 4000, dual AHRS 3000S 1454A2-8006.</td>
<td></td>
<td></td>
<td>EASA TCDS IM.A.207 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Avionics Design Services Ltd. Master Drawing List MDL09001 Issue 1, Revision B from 29 July 2011</td>
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<td>10036054</td>
<td>BOMBARDIER INC.</td>
<td>TC1A STC: SA04-113</td>
<td>REV. 1</td>
<td>TCC A STC: SA04-113</td>
<td>BD-700-1A10</td>
<td>Installation: Complete Custom Aircraft Interior - ASN 9406</td>
<td>EASA IM.A.090 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>TCCA STC: SAA4-113 Issue 1 dated 28 October 2004</td>
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<tr>
<td>10036058</td>
<td>BOMBARDIER INC.</td>
<td>TC1A STC: SA05-80</td>
<td>REV. 1</td>
<td>Complete Custom Aircraft Interior - ASN 9421</td>
<td>BD-700-1A11</td>
<td>Installation: Complete Custom Aircraft Interior - ASN 9421</td>
<td>EASA IM.A.090 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>TCCA STC: SAA5-80 Issue 1 dated 14 June 2005</td>
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<td>10036059</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC SA738NM</td>
<td>B777-200</td>
<td>Installation of an extended upper deck crew rest in accordance with flight structures Inc. Master Drawing List 90SF004-001, Revision E dated July 13, 1992, or later FAA approved revisions. Validation of FAA STC SA738NM...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>None.</td>
<td>None.</td>
<td>10/08/2011</td>
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<td>10036059</td>
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<tr>
<td>10036080</td>
<td>BOMBARDIER INC.</td>
<td>REV. 1 TCCA STC SA4-113</td>
<td>BD-700-1A10</td>
<td>Installation of cabin systems and custom furnishings applicable to BD-700-1A10 serial number 9393</td>
<td>EASA M.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The certification requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. SA4A-113 Issue 1 dated 28 October 2004.</td>
<td>Restricted to ASN 90391. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOMS. 2.WLAN system is compatible only to IEEE 802.11b.</td>
<td>09/08/2011</td>
<td>Active</td>
<td>10036080</td>
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<tr>
<td>10036081</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>FAA STC ST10897SC-D</td>
<td>B737-700</td>
<td>VIP seat change / Replacement of single seat by one with massage feature.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List MDL-AY01-2520 rev. B dated 20.01.2011 or later revisions of the above listed documents approved by EASA.</td>
<td>This approval is limited to Aircraft B737-700GW, Serial Number 36027.</td>
<td>09/08/2011</td>
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<td>10036081</td>
<td>ASSOCIATED AIR CENTER, L.P.</td>
<td>REV. 2 FAA STC ST10897SC-D</td>
<td>B737-700</td>
<td>Change to EASA STC 0010038081 (Installation of an Executive interior on B737-700GW aircraft. Project No.:AAC-06-08-05) Installation of the NEDA Model 4402 Single occupant Side facing divan.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Plan, Reference 299C000090, initial revision, dated 27 October 2011. Airplane Flight Manual Supplement, Reference 2582010000, Revision A dated 12 November 2010. Master Data List, Reference 2582050090, Revision C, dated 19 April 2012. Instructions for Continued Airworthiness, Reference 2582050090, Revision D, dated 31 January 2012.</td>
<td>1) This approval is limited to Aircraft B737-700, Serial Number 36027.</td>
<td>29/06/2012</td>
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<tr>
<td>10036080</td>
<td>INNOTECH AVIATIONA DIVISION OF IMP GROUP LTD.</td>
<td>FAA STC ST10897SC-D</td>
<td>B737-700</td>
<td>ICG 0700 Satcom Upgrade.</td>
<td>EASA M.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The certification requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. O-LSA07-338D Issue 1 dated 19 December 2007. Innotech Aviation, a Division of IMP Group Ltd. Master Drawing List 08LZ.0X-0528-126 Revision N/C dated 18 December 2007, prior approved revision. Maintenance must be in accordance with Innotech Aviation, a Division of IMP Group Ltd. Instruction...</td>
<td>Restricted to AS 9026.</td>
<td>10/08/2011</td>
<td>Active</td>
<td>10036080</td>
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<td>10036091</td>
<td>INNOTECH AVIATIONA DIVISION OF IMP GROUP LTD.</td>
<td>FAA STC ST10897SC-D</td>
<td>B737-700-1A10</td>
<td>Software and DNL Upgrade to Existing HSD-400 Satcom Installation.</td>
<td>EASA M.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The certification requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. O-LSA08-197D Issue 1 dated 29 August 2008. Avionics Design Services Ltd. Master Drawing List MDL0888 Revision N/C dated 29 August 2008 or later approved revision.</td>
<td>Restricted to AS 9026.</td>
<td>10/08/2011</td>
<td>Active</td>
<td>10036091</td>
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<tr>
<td>BOMBARDIER INC.</td>
<td>BD-700-1A10</td>
<td>Interior Modifications - ASN 9020 - Installation, Modifications to APU Bay Light, Galley Oven, Smoke Detector, Wiring and Equipment Bay Provisions</td>
<td>EASA MA.009</td>
<td>Transport Canada STC No C-LSA01-452/D</td>
<td>Restricted to ASN 9020</td>
<td>10/06/2011</td>
<td>Active</td>
<td>100360095</td>
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<td>BOMBARDIER INC.</td>
<td>BD-700-1A10</td>
<td>Miscellaneous Cabin Modifications</td>
<td>EASA MA.009</td>
<td>Transport Canada STC No C-LSA02-065/D</td>
<td>Restricted to ASN 9020</td>
<td>10/06/2011</td>
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<td>ACS-NAI Ltd.</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Models</td>
<td>EASA MA.009</td>
<td>TCCA STC No C-LSA11-134/D</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td>Installation of PFD Aspen EFG1000 PRO</td>
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<td>Flight Manual Supplement</td>
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<td>BINDER MOTORENBAU GmbH</td>
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<td></td>
<td>Design: Non-significant change. The Certification Basis of the original product remains applicable. Additional/ Replacement: The associated certified noise and/or emissions levels of the original product are unchanged. Additional/ Replacement: The associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Binder Motorenbau STC &quot;vita Umrüstung SKB&quot; 12.02.2011/Beschreibung der Umrüsten und Zeichnung etc 2.03/Erlassungen des Flughandbuches betreffend STC &quot;Umrüstung SKB&quot; für den Motorsegler etc.</td>
<td>None.</td>
<td>11/02/2011</td>
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<td>AIRWOLF AEROSPACE L.L.C.</td>
<td>REV. 1</td>
<td>FAA STC SR02646Ch</td>
<td>TENSION-TORSION STRAP - REPLACEMENT OF LIFE-LIMITED PARTS - Amended Airworthiness Limitation Section of the Instruction for Continued Airworthiness AA-206-ICA</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airworthiness Limitation Section of the Instruction for Continued Airworthiness AA-206-ICA Rev. E, FAA Approved on 19 November 2012 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/02/2013</td>
<td>Active</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>B747-400 SERIES</td>
<td>Replacement of Global Positioning System Sensor Unit and GPS Antenna.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Lufthansa Suplemental Type Definition Document STDO-105SM Issue 1, Rev. 5, dated Aug. 11, 2011. Compliance Record LufTRA-61-0005 SM Iss01 Rev0 dated 11 August 2011. LufTRA-61-0005 SM Iss01 Rev R dated 25 March 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>19/08/2011</td>
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<td>CLIPPER AVIATION, Inc.</td>
<td></td>
<td>FAA STC S70231NA</td>
<td>Installation of seating configuration on Upper Deck of B747-400 Boeing Converted Special Freighter</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List CLJ-2006-053, Revision D dated May 23, 2011. Airplane Flight Manual Supplement CLJ-2007-015 Rev. Ifor later revisions of the above listed documents approved by EASA. The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>None.</td>
<td>12/06/2011</td>
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<tr>
<td>AIRBUS engineering GmbH</td>
<td>REV. 1</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Honeywell EGPWS MK VI System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - The following paragraph(s) at a later amendment: CS-23 Amendment 02: The requirements for environmental protection and the associated certified noise and/or emissions lev. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - The following paragraph(s) at a later amendment: CS-23 Amendment 02: The requirements for environmental protection and the associated certified noise and/or emissions lev...</td>
<td>- Flight Manual Supplement PO-1106-34000-0000 Engineering Work Order: E9/1106-34000-0000 Description of Modification: DE-1106-00000-0000 Maintenance Manual Supplement MM-1106-00000-0000 for the above listed document(s) approved/ accepted under the EASA system.</td>
<td>The aircraft must have been already equipped with a suitable display and suitable sensors. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon...</td>
<td>10/02/2021</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td></td>
<td>SR20, SR22</td>
<td>Installation of a Primary Function Display System EXP9400</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23, Amendment2: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23, Amendment 2: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23, Amendment 2: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23, Amendment 2: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23, Amendment 2: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23, Amendment 2: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and the following additional or alternative</td>
<td>Flight Manual Supplement: ASR-2011-028-FMS-05-000-05 for the above listed documents approved by EASA. Instructions for Installation: ASR-2011-028-IF1-10-00 Instructions for Continued Airworthiness: ASR-2011-028-ICA-11-00</td>
<td>The aircraft must have been already equipped with suitable GPS system.</td>
<td>16/06/2011</td>
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List of EASA Supplemental Type Certificates

<table>
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<th>STC Holder</th>
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<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>AIRLIFT ASI</td>
<td>A31328</td>
<td>AS350B3</td>
<td>AS350B3, AS350B1</td>
<td>Installation of Bendix/King KJA-45SB Radar Altimeter.</td>
<td>The Certification Basis for the original product and the following/additional or alternative avionics/requirement requirements are applicable to both certificates/approval. Requirements of CS-27 at amdt. 2 are complied with by the modification. The requirements for environmental protection and the associated technical documentation.</td>
<td>Airlift ASI Master Document List MDL-130-34-100 revision 0 dated 08.08.2011.</td>
<td></td>
<td>17/06/2011</td>
<td>Active</td>
<td>10036162</td>
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<tr>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>B200C2/B200CT/B200G</td>
<td>TBE200T A200CT (RC-12H), B200/B200C A200CT (RC-12G) A200CT (FWC-12D) A200CT (C-12F), A200CT (RC-12H) A200C (C-12C), A200CT (C-12D)</td>
<td>Installation of 5 blade MTV-271-E-C-F-R (P) CFR225-55F propellers.</td>
<td></td>
<td>The Certification Basis for the original product and the following/additional or alternative avionics/requirement requirements are applicable to both certificates/approval.</td>
<td>E-2041, Airplane Flight Manual Supplement, revision 0, dated 21.06.2011E-2042, Installation Instruction, revision 0, dated 21.06.2011E-2043, Instructions for Continued Airworthiness, Revision 0, dated 01.06.2011 for later revisions of the above listed documents approved by EASA</td>
<td></td>
<td>none</td>
<td>19/08/2011</td>
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<td>INNOVATIVE SOLUTIONS AND SUPPORT, I</td>
<td>B767-200/300/300F</td>
<td>FAA STC ST0160NY</td>
<td>Installation of Flat Panel Display System</td>
<td></td>
<td>Installation Configuration: 6</td>
<td>Installation Configuration 6 approved only. The Smiths Aerospace Limited Integrated Standby Instrument System (ISIS) is not approved for installation under this STC. No earlier revisions of the listed documents are approved under this STC.</td>
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<td>23/08/2011</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>501, 551</td>
<td>Installation of a TADS System L3 Communication SKY899</td>
<td>The Certification Basis for the original product and the following/additional or alternative avionics/requirement requirements are applicable to both certificates/approval.</td>
<td>Flight Manual Supplement: ASR-2011-009-FMS-05-00 Rev. 00 for later revisions of the above listed documents approved by EASA</td>
<td>The aircraft must have been already equipped with a suitable Multifunction Display.</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>208, 208B</td>
<td>Installation of a Satellite Antenna Chelton HGA-7000</td>
<td>The Certification Basis for the original product and the following/additional or alternative avionics/requirement requirements are applicable to both certificates/approval.</td>
<td>Flight Manual Supplement: ASR-2011-012-FMS-05-00 Rev. 00 for later revisions of the above listed documents approved by EASA</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>A340-300</td>
<td>Omnibus Modification for the Approval of Air New Zealand Engineering Orders Embodied on the A/C</td>
<td>The Certification Basis for the original product remains applicable to the aircraft. The requirements for environmental protection and the associated certificates of noise and/or emissions levels of the original product remain applicable to this approval.</td>
<td>Master Document List, 521- MDL-0810, Issue 1, 15/07/2011 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>24/08/2011</td>
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<td>TELAIR INTERNATIONAL, LLC</td>
<td>B747-8F</td>
<td>Lower Deck Cargo Handling System, Installation</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the aircraft. EASA CRI A-100. The requirements for environmental protection and the associated certificates of noise and/or emissions levels of the original product remain applicable.</td>
<td>PSCP 193100CPP-8F and FAA STC 02365LA or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>26/08/2011</td>
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<td>TELAIR INTERNATIONAL, LLC</td>
<td>B747-8F</td>
<td>Main Deck Cargo Handling System, Installation</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the aircraft. EASA CRI A-100. The requirements for environmental protection and the associated certificates of noise and/or emissions levels of the original product remain applicable.</td>
<td>PSCP 193100CPP-8F and FAA STC 02365LA or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>26/08/2011</td>
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<td>AIRBUS S.A.S.</td>
<td>Airbus Corporate Jet Centre Modification CJ1000 Issue 1 - VIP Interior - &quot;RAFO 3&quot;.</td>
<td>In accordance with the EASA TCDS, The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the aircraft. EASA CRI A-100. The requirements for environmental protection and the associated certificates of noise and/or emissions levels of the original product remain applicable.</td>
<td>- Modification Approval Sheet, ref: CJ1000, Issue 1, dated 23rd August 2011 - AFM Supplement, ref: FM- S MD-1000, Revision 00, dated August 2011 - Conversion Instruction, ref.: CI C1000-07, Revision A, dated 25th July 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>This STC is limited to A320-214 MSN 4795.</td>
<td>24/08/2011</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>BD 100-1A10 (CHALLENGER 300)</td>
<td>Installation of EGPS MK-V with Runway Awareness and Advisory System (RAAS).</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the aircraft. EASA CRI-RAAS Issue 3 dated 16/08/2011. The requirements for environmental protection and the associated certificates of noise and/or emissions levels of the original product remain applicable.</td>
<td>- Master Document List (MSL) 338A-MDL, dated 22/08/2011 - Airplane Flight Manual Supplement 326A - FMS dated 5/08/2011 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>25/08/2011</td>
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<td>SABENA TECHNICS BQC</td>
<td>A340-311/312-313</td>
<td>A340 Cabin Reconfiguration &amp; Zodiac Flight Entertainment Installation</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the aircraft. EASA CRI-RAAS Issue 3 dated 16/08/2011. The requirements for environmental protection and the associated certificates of noise and/or emissions levels of the original product remain applicable.</td>
<td>Master Data List, ref: 7048-01-A-21, R02 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>09/08/2012</td>
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<td>S.E.A.M. SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
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<td>B767-300</td>
<td>Astronics EMPower 110VAC</td>
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<td>AMES - AEROSPACE AND MECHANICAL ENGINEERING SERVICES</td>
<td>767-300</td>
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<td>SERRA INDUSTRIES LTD.</td>
<td>FAA STC</td>
<td>ST0236LA</td>
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<td>HAWKER 800XP</td>
<td>Installation of Honeywell TFE731-50R Engines.</td>
<td>Certification Basis</td>
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<td>AEROLITE AG</td>
<td>REV. 2</td>
<td>AB 139, AW 139</td>
<td>10036324</td>
<td>EMS Equipment PIN 139250-501</td>
<td>Installation of Honeywell TFE731-50R Engines.</td>
<td>Certification Basis</td>
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<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA STC</td>
<td>SAU1946LA</td>
<td>10036326</td>
<td>PA-31T, PA-31T1, PA31T2</td>
<td>Installation of Powerplant Indication System.</td>
<td>Certification Basis</td>
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<td>10036328</td>
<td>CONTINENTAL AEROSPACE TECHNOLOGIES</td>
<td>REV. 4</td>
<td>DJ 40 D</td>
<td>DA 40 D</td>
<td>Installation of engine TAE 125-02-114 in combination with propeller MTV-6-A/187-129 or MTV 6-A/100-89R Revision 1 follows a change of ownership. Revision 2 includes propeller MTV 6-A/195-69.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI A-01 for Diamond DA40, Issue 21, dated 22.09.2008; Equivalent Safety Finding (s): CRI A-01 for Diamond DA40, Issue 21, dated 22.09.2008. This certification is applicable to this certificate/approval.</td>
<td>- Approved Flight Manual Supplement for the DA 40 D with TAE 125-02-114 engine installation. TAE Nr.: 56-0310-50022, Issue 2. Revision 0- Anhang zum Flughandbuch Daimond DA 40 D mit TAE 125-02-114 installation. TAE-Nr.: 50-0310-50012, Ausgabe 3, Revision 0 or later. EASA approved issues/Revisions- Type D...</td>
<td>21/01/2020</td>
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<td>10036331</td>
<td>DONALDSON COMPANY, Inc.</td>
<td>REV. 3</td>
<td>FAA STC SR02338CH</td>
<td>A119, AW119MKII</td>
<td>Installation of a foreign object damage (FOD) screen or an inlet barrier filter (IBF) system Revision 1 reflects the name and address change from Aerospace Filtration Systems, Inc. to Donaldson Company, Inc. Revision 2 updates the Associated Technical Documentation with regards to “later revisions.”</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>1. Rotorcraft Flight Manual Supplement is the document AFS-AA119EASA-BF-FMS Revision R2. Air Filtration System Inc. Master Data List Document No: AFS-AA119-BF-MDL. Revision 1 or later revisions of the above listed document(s) approved/accepted condition of the certification</td>
<td>None</td>
<td>10/11/2021</td>
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<td>10036344</td>
<td>BOMBARDIER INC.</td>
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<td>TCCA STC NO. 5A1-11-69</td>
<td>BD-700-1A10, -1A11</td>
<td>Installation of a foreign object damage (FOD) screen or an inlet barrier filter (IBF) system Revision 1 reflects the name and address change from Aerospace Filtration Systems, Inc. to Donaldson Company, Inc. Revision 2 updates the Associated Technical Documentation with regards to “later revisions.”</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>1. Rotorcraft Flight Manual Supplement is the document AFS-AA119EASA-BF-FMS Revision R2. Air Filtration System Inc. Master Data List Document No: AFS-AA119-BF-MDL. Revision 1 or later revisions of the above listed document(s) approved/accepted condition of the certification</td>
<td>None</td>
<td>02/09/2011</td>
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<td>10036346</td>
<td>ACS-NAI Ltd.</td>
<td></td>
<td>TCCA STC C-LSA11-05860</td>
<td>BD-700-1A10</td>
<td>Installation of Tailwind 500 System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC No. SAS11-49 Issue 1 dated 08 August 2011. ACS-NAI Ltd. Modification Data Summary G733000 Revision NC or later approved revision. Bombardier Instruction for Continued Airworthiness G733000-ICA Revision R3 or later revision. Bombardier Supplemental Airworthiness Limitations, G733000-A.</td>
<td>None</td>
<td>02/09/2011</td>
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<td>10036347</td>
<td>INMOTEC AVIATION DIVISION OF IMP GROUP LTD.</td>
<td></td>
<td>TCCA STC C-LSA11-07960</td>
<td>BD-700-1A10</td>
<td>Installation of Complete Custom Aircraft Interior - ASN 9399</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- Transport Canada STC No. C-LSA11-05860/D Issue 1 dated 21 April 2011. ACS-NAI Ltd. Modification Data Summary H505000 Revis 0-C or later approved revision. - ACS-NAI Ltd. Instruction for Continued Airworthiness H505000-ICA Revision R7 or later approved revision. - Transport Canada STC No. C-LSA11-05860/D Issue 1 dated...</td>
<td>None</td>
<td>02/09/2011</td>
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<td>EASA STC no.</td>
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<td>10036349</td>
<td>ADAC LUFTHANFT TECHNIK GmbH</td>
<td>REV. 2</td>
<td></td>
<td>EC135 P3, EC135 T3, EC135 P2+, EC135 T2+</td>
<td>Installation of an NVIS lighting system in an Airbus Helicopter EC135 (MEGHA5) Note 1. Revision 1 of the STC introduces a filter on the display of the NVG30 units (dual installation), upgrades the SAT phone control (aeroplane) and corresponding software upgrade of the DMAP System and deactivates the...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:The following amendment: CS 27 Amendment 3 as per Doc EC135-05162-027- RCS-02, Rev. 2. The requirements for environmental protection and the associate...</td>
<td>STC 10036349 (Applicable to EC135 T2+EC135 P2+ to EC135 T2(CP05) - EC135P2(CP05)) - EMS EC135-01663-025-MFS. Revision STC 19003649 Revision 1 (Applicable to EC135 T2+ - EMS EC135-02565-025-MFS. Revision STC 19003649 Revision 2 (Applicable to EC135 P2+ to EC135 T2+ to EC135 T3 (CP05), EC135 P3...</td>
<td>02/12/2016</td>
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<td>10036361</td>
<td>LEES AVIONICS Ltd.</td>
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<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>B200CTG, B200CT, B200T A200C, A200CT, B200C 200, 200C, 200CT, 200T, A200 1900, 1900C, 1900D</td>
<td>TAWS-B Installation</td>
<td>CS-23 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data LIMDI 78901 Issue 2 dated 08 December 2012: Change Statement CS-778 Issue 2 dated 08 December 2012: Instruction: Continued Airworthiness ICA-778 Issue 1 dated 30 April 10 Flight Manual Supplement AFS 778-01 Issue 1 dated 30 April 2012, or later revisions of the above listed documents approved...</td>
<td>None</td>
<td>02/09/2011</td>
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<td>10036367</td>
<td>S.E.A.M SERVICE ELECTRONIC AVION MARINES.A.R.L</td>
<td></td>
<td></td>
<td>F90</td>
<td>STC 2598/09 Rev 4 Dual EFIS G100 installation</td>
<td>The Certification Basis for the original product remains applicable to the certificates/approval. The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Dossier 2598/09 Rev 4 dated 18 July 2011: Supplement manual de vol SEAM/SMV/2598/09 Rev 4 dated 18 July 2011: or later revisions of the above listed documents approved by EASA.</td>
<td>Limitation to Beech F90 equipped with SP2000 autopilot and modified according to SEAM STC 2465/08 avionics installation.</td>
<td>02/09/2011</td>
<td>Active</td>
<td>10036367</td>
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<td>10036381</td>
<td>MARSHALL OF CAMBRIDGE AEROSPACE-LIMI TED</td>
<td></td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>B737-700</td>
<td>737 Medvac Role Change</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements applicable to the certificates/approval: CS024 Amdt 6 CRD-0115/000036 Medical Evacuation The certified noise and/or emissions levels of the original product unchanged and...</td>
<td>Certificate of Design CS/ B737-1707/010563 Issue 1 dated 25/05/2011: AFM Supplement AFS/ 737MED(1797)/13/0011 Rev 0 dated Jan 2011: or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>06/06/2011</td>
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<td>10036381</td>
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<td>10036393</td>
<td>HELI-ONE (NORWAY) AS</td>
<td>REV. 1</td>
<td></td>
<td>AS 332 L, AS 332 L1 AS332 L1L2</td>
<td>Installation of STAR SAFIRE FLIR system consisting of external mount, internal and external cabling and installation of mission console in Cabin. Revision 0 – initial approval applicable to AS332 L1L2. Revision 1 – STC applicable corrected</td>
<td>The Certification Basis for the original product remains applicable to the certificates/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact...</td>
<td>AHS-124 dated 24 May 2011: Flight Manual Supplement FLIR STAR SAFIRE/HARDWARE-12556 revision C dated 30 August 2011: Compliance Checklist - 12556 revision A dated 24 May 2011: Continued Airworthiness/Supplement for FLIR HD Installation Certification Plan revision (--) dated 17 December 2008: HSE-567...</td>
<td>Flight into known icing Conditions prohibited.Prior to installation of this change it must be determined that the relationship between this change and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product...</td>
<td>01/02/2024</td>
<td>Active</td>
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<td>10036411</td>
<td>SPECTRUM AEROMED, LLC/ B SPECTRUM AEROMED</td>
<td>REV. 2</td>
<td></td>
<td>FAA STC SAI1213CH</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Install air ambulance Note 1. This STC reflects the transfer STC CAA ES S937-1 from SpectrumAeromed, Inc. to Smiley Creek Enterprises, LLC. The STC superseded CAA ES S937-1. EASA STC CAA STC 10036413 superseded EASA STC CAA IM.A.S.C. 01613. This STC, together with FAA STC SAI1213CH and 10036413 superseded EASA STC CAA IM.A.S.C. 024...</td>
<td>The Certification Basis for the original product remains applicable to the certificates/approval, except where amended by additional or later amendments if instituted on FAA STC SAI1213CH. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>FAA Approved Model List (AML) SAI1213CH, released 28 February 2019, greater FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse affect upon the airworthiness of the product whose STC is valid for aircraft types and models a...</td>
<td>24/04/2019</td>
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List of EASA Supplemental Type Certificates

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<th>STC No.</th>
<th>STC Holder</th>
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<tr>
<td>10036412</td>
<td>SPECTRUM AEROMED, LLC/ b/a SPECTRUM AEROMED</td>
<td>FAA STC SA1/1213CH</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Install an ambulance conversionNote:1. This STC reflects the transfer of STC CAA ES S3741 from SpectrumAeromed, Inc. to Snaky Creek Enterprises LLC.2. This STC supercedes STC CAA ES S3741.ESA STC IM.A/3.C.I.15/18.1. This STC, together with EASA STCs 10036411 and 10036413, supersedeEASA STC IM.A...</td>
<td>FAA Approved Model List (AML) No. SAA1213CH, dated 28 February 2019, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a functional effect upon the airworthiness of the aircraft.</td>
<td>24/04/2019</td>
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<tr>
<td>10036413</td>
<td>SPECTRUM AEROMED, LLC/ b/a SPECTRUM AEROMED</td>
<td>FAA STC SA1/1213CH</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of an ambulance conversionNote:1. STC 10036413 reflects the transfer of STC CAA ES S374-1 from SpectrumAeromed, Inc to Snaky Creek Enterprises, LLC. STC 10036413 supersedeSTC ESA-STC IM.A/3.C.I.15/18.1. STC 10036413, together with EASA STCs 10036411 and 10036412, supersedeEASA STC IM.A/2..</td>
<td>See EASA Approved Model List prior revisions of the above noted STCs approved by EASA in accordance with EASA Decision 2004/54/EC (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/ US bilateral Agreements.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a functional effect upon the airworthiness of the aircraft.</td>
<td>24/04/2019</td>
<td>Active</td>
<td>10036413</td>
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<tr>
<td>10036439</td>
<td>AVCON INDUSTRIES, Inc.</td>
<td>FAA STC SA1/0415/WH</td>
<td>B90, C190, C90A, E90, H90 300, 300LH, B300 200, 200T, A200, B200, B200T 100, A100, A100A, A1100, B110</td>
<td>Installation of Baggage Door.</td>
<td>The Certification Basis for the original product includes applicable ETSI certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Avon Industries Inc, Master Drawing List No 3D500, Revision C, dated 27 March 1996. Instructions for Continued Amortization of EASA STCs approved by EASA.</td>
<td>06/02/2011</td>
<td>None</td>
<td>10036439</td>
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<tr>
<td>10036443</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>P.68TC &quot;OBSERVER&quot; P.68 &quot;OBSERVER 2&quot; P.68 &quot;OBSERVER&quot;</td>
<td>Installation of different mission equipment in a Boeing P.68 &quot;OBSERVER&quot; aircraft. The aircraft cabin has to be modified and provisions for accommodation of Mission Equipment have to be installed. The installation omission equipment itself is not part of this STC...</td>
<td>Certification basis as per EASA TCDS: FAR 23, applicable Amendment: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: Applicant selected to comply CS-23, Amendment 2. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Installed in accordance with EASA Decision 2004/54/EC (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/US bilateral Agreements.</td>
<td>06/02/2011</td>
<td>None</td>
<td>10036443</td>
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<td>10036450</td>
<td>SCANDINAVIAN AVIONICS DESIGN A/S</td>
<td>GULFSTREAM 200 / GALAXY</td>
<td>UPGRADE OF ROCKWELL COLLINS TDR-94D MOD E TRANSPONDER SYSTEM FORENHANCED (EHS) SURVEILLANCE CAPABILITY.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: Certification basis as per EASA TCDS: FAR 23, applicable Amendment: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. Applicant selected to comply CS-23, Amendment 2. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>To be installed in accordance with Engineering Order Doc. ID 2006103-ED, Issue 1 and Technical Order Doc. ID 2006103 Issue 1To be operated in accordance with Airplane Flight Manual SupplementDoc. ID 200613-AM, Issue 1To be maintained in accordance with Continued Amortization Doc. ID 200613-CA, Issue 1...</td>
<td>06/02/2011</td>
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<td>STAR AVIATION, Inc.</td>
<td>FAA STC 10036456</td>
<td>B747-400</td>
<td>Installation of Connexion by Boeing Next Generation Avionics Data System in accordance with Star Aviation Inc. Master Data ListR44-0163, Revision 0, Dated May 12, 2011 or later approved</td>
<td>Aircraft Flight Manual Supplement R44-0300 Rev. IR dated October 16, 2004 for Continued Airworthiness Star Avionics Document NoR44-0273, Rev. D, dated June 20, 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/4/C (or subsequent).</td>
<td>None</td>
<td>09/02/2011</td>
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<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA 10036469</td>
<td>CESSNA 208B</td>
<td>Replacement of the original P=+4/½W PT6A-114(A) engine and propeller with P=+4/½W PT6A-424 and 4-blade Hartzell PropellerH-804-3P0/30900(K) and associated equipment in accordance with FAA STC SA2357LA.</td>
<td>Master Drawing List 200830-000, Revision IR, dated June 8, 2011. Flight Manual Supplement AMFS 200830, Rev. IR, dated June 22, 2011. Instruction of Continued Airworthiness ICA 200813-30, Rev. A, dated May 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/4/C (or subsequent).</td>
<td>1. AeroAcoustics Aircraft Systems, Inc. “Aircraft Payload Extender STOL System”, STC SA01805SE (EASA STC No. 1002821) is required as part of this alteration. 2. This modification is not compatible with Cessna Model 208B Serial Numbers 20881190, 20881216, and 20882000 and greater (Garmin 1000 installed...</td>
<td>12/09/2011</td>
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<td>AERODATA AG</td>
<td>FAA 10036478</td>
<td>LBA EMZ NR. TA6007</td>
<td>Change to LBA STC approval “EMZ Nr. TA6007” AFM approvalDTM-721A” dated 26 June 2002. Change of procedures “Instructions for Continued Airworthiness” in accordance with RVSM qualification and update of the aircraft configuration.</td>
<td>Master Data List doc; N° MDL-AC-047, Rev. 5 dated September 06 2011. Instructions for Initial and Continued Airworthiness doc; N°HB-A/ C-650, Rev. 4 dated September 06 2011. Airplane Flight Manual Supplement doc; N° FMS-047, Rev. 2 dated August 02 2011 or later revisions of the above listed documents ap...</td>
<td>Restricted to SN 183</td>
<td>12/09/2011</td>
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<td>AERODATA AG</td>
<td>FAA 10036479</td>
<td>LBA EMZ NR. TA6000</td>
<td>Change to LBA STC approval “EMZ Nr. TA6000” AFM approvalDTM-721A” dated 06 June 2002. Change of procedures “Instructions for Continued Airworthiness” in accordance with RVSM qualification and update of the aircraft configuration.</td>
<td>Master Data List doc; N° MDL-AC-044, Rev. 3 dated September 06 2011. Instructions for Initial and Continued Airworthiness doc; N°HB-A/ C-645, Rev. 2 dated September 2011. Airplane Flight Manual Supplement doc; N° FMS-046, Rev. 3 dated 02 August 2011 or later revisions of the above listed documents ap...</td>
<td>Restricted to SN 216</td>
<td>12/09/2011</td>
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<td>10036482</td>
<td>LEES AVIONICS Ltd.</td>
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<td>Installation of the Spectrum search light STX16</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable for the STC approval. The Applicant applied as voluntary compliance the CS 29-2 for the applicable paragraph C.201(9).</td>
<td>None.</td>
<td>16/02/2011</td>
<td>Active</td>
<td>1032942</td>
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<td>10036485</td>
<td>SKYTRAC SYSTEMS Ltd.</td>
<td>REV. 1</td>
<td>TCSC STC H10-24</td>
<td>AS 350 F2, AS 350 N</td>
<td>Sky Track STC AIRBORNE POSITION / COMMUNICATION UPGRADE Modification upgrades an existing ISAT-100 Airborne Position / DataCommunicator system installation to an ISAT-200 system. This EASA approval validates the TCSC STC H10-24 (issue 3 on the product models referenced above). Revision 1 is an administrative change.</td>
<td>- Flight Manual Supplement FMS10033-5: Revision N/C dated March 14, 2011: Avionics Design Services LTD Master Drawing List MDL10033, Revision D, Issue 1, dated March 1, 2011 (relative to the EC AS 350 F, F1, F2). - Avionics Design Services LTD Drawing List DL10033-5: Revision N/C, Issue 1, dated M.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not result in a new adverse effect upon the airworthiness of the product.</td>
<td>29/05/2024</td>
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<td>10036488</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td></td>
<td>TCSC STC NO. SAT1-37</td>
<td>DHC-8-401 -402</td>
<td>Temporary Removal of Auxiliary Power Unit (APU)</td>
<td>The temporary APU removal series must be done in accordance with TCSC Approved Field Aviation Company Inc. Modification Instruction 25341000, Revision A, dated 02 June 2011.</td>
<td>- This STC reference specified interfacing equipment that is approved within the data. The ISAT-200 is capable of being interfaced to additional aircraft systems for which approval is beyond the scope of this STC. Such interfaces may need further approval.</td>
<td>13/09/2011</td>
<td>Active</td>
<td>10328448</td>
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<tr>
<td>10036493</td>
<td>AIM AVIATION (HENSHALLS) LTD</td>
<td></td>
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<td></td>
<td>Installation of a Galley 13 and Closet C’tins</td>
<td>CS 25 Amendment 11</td>
<td>Change Record HEN 1208 Issue 1a: for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>13/09/2011</td>
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<td>10036501</td>
<td>AIRCRAFT DESIGN CERTIFICATION GmbH</td>
<td>REV. 1</td>
<td>EA 300L, EA 300LC</td>
<td>EA 300 300/3</td>
<td>Multi glider towing capacity up to 1000kg total tow mass</td>
<td>The Certification Basis for the original product remains applicable to this STC approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List ADc-44-001M1, dated August 18, 2011. Airplane Flight Manual Supplement ADc-44-001M1, dated August 30, 2011.</td>
<td>None</td>
<td>17/07/2018</td>
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<td>Certification basis</td>
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<td>10036506</td>
<td>LEES AVIONICS Ltd.</td>
<td>Active</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BELL 412</td>
<td>Installation of the UNS-1E FMS.</td>
<td>The Certification Basis for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Rotorcraft Flight Manual Supplement RFMS-1040 Issue 1 &amp; MDL 1041-01 Issue 9.</td>
<td>This kit/STC can be installed on helicopter that has installed the CRI H-1 EWIS ICA. EASA approved EASA R.S.01465.</td>
<td>14/02/2011</td>
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<td>10036508</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td>Active</td>
<td>AIW1005P</td>
<td>HAWKER 800XP</td>
<td>NAV 434 “Installation of Helicopter Emergency Egress Light (HEEL)”</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Modification Bulletin NAV-434 Rev. 0</td>
<td>None.</td>
<td>14/02/2011</td>
<td>Active</td>
<td>10036508</td>
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<td>10036511</td>
<td>32B SUPPORT SERVICES GmbH</td>
<td>REV. 3</td>
<td>HAWKER 800XP</td>
<td>HAWKER 800XP</td>
<td>Installation of a Flight Data Recorder (FDR) System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with 32B Design GmbH EngineeringOrder ref AM1581009A-01 dated 10.12.2014. In accordance with FAA STC Approval. The original product remains unchanged and is applicable to this certificate/ approval.</td>
<td>None.</td>
<td>27/09/2018</td>
<td>Active</td>
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<td>10036542</td>
<td>B/E AEROSPACE, Inc.</td>
<td>REV.</td>
<td>FAA STC ST02871CH</td>
<td>B777F, B777-300, B777-300ER B777-200, B777-200LR</td>
<td>Installation of B/E Aerospace, Inc., Vacuum Toilet Assembly Retrofit Kit.</td>
<td>The Certification Basis for the original product and the following additional/ alternative airworthiness requirements are applicable to this certificate/ approval. CRI H-1 EWIS ICA.</td>
<td>Installation Instructions AAM-38-3003 Rev. 3 dated 10 May 2011 Instructions for Continued airworthiness of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>19/06/2011</td>
<td>Active</td>
<td>10036542</td>
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<td>10036545</td>
<td>FOKKER SERVICES B.V.</td>
<td>REV. 2</td>
<td>777-200/-200LR/300/-300 ER/-F</td>
<td>Installation of a GKN Aerospace designed Boeing 777 No. 1 Windsail/Part No. 10702 (LH) / 10703 (RH); or Part No. 10712 (LH) / 10713 (RH) Revision 2. Transfer from GKN to Fokker Services.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Replacement of a Flight Compartment Window Number 1 per this STC GKN Aircraft Engineering Order AM1581009A-01 Issue 1 &amp; MDL 1041-01 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>17/07/2017</td>
<td>Active</td>
<td>10036545</td>
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<td>10036548</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST02869WY-O</td>
<td>A330-341, -342, -343 A330-321, -322, -323 A330-301, -302, -303</td>
<td>Provision of a Panasonic Ex Phone System (Validation of FAA STC ST02869WY-O). This STC installs provisions (antenna and cabling) for Panasonic Ex Phone System. The Revision 1 of this STC corrects the applicability by editing helicopter models, omitted at the initial approval.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List, Ref: 0154-0760-2234, dated 08 September 2011, FAA STC Approval. Ref: 0154-0765-2234 Revision of later revisions of the above listed documents approved by EASA.</td>
<td>This STC approval only certifies the provisions for the Panasonic ExPhone. This device activation, for which these provisions are intended, will be certified separately.</td>
<td>10/11/2011</td>
<td>Active</td>
<td>10036548</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10036558</td>
<td>B/E AEROSPACE, Inc.</td>
<td>FAA STC ST02876CH</td>
<td>B737-600/700/800/900</td>
<td>Installation of Vacuum Toilet Assembly Retrofit Ki P/N 4100390-003.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/ approval: CFR-h.1 EWIS ICA.The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td></td>
<td>None.</td>
<td>10/02/2011</td>
<td>Active</td>
<td>10030535</td>
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<tr>
<td>10036586</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>LEARJET 35A</td>
<td>Installation of Honeywell Allied Signal Enhanced Ground Proximity Warning MK VIII System</td>
<td>Certification Programme 105-LR Rev IR dated June 21, 2011 and Engineering Order No.: 105-LR, 088.01 Rev 5 dated June 21, 2011 for later revisions of the above listed documents approved by EASA</td>
<td></td>
<td>None</td>
<td>21/09/2011</td>
<td>Active</td>
<td>10030536</td>
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<tr>
<td>10036587</td>
<td>AVIONIK ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td>2. PA-31T, PA-31T1, 2. PA-31T2, PA-31T3 2. PA-31P, PA-31P-350, 1. PA-31, PA-31-300, 1. PA-31-320, PA-31-350, Installation of one or more Avionik Straining Heater Control Units/HC200.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23 Amendment 2The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Flight Manual Supplement ASR-2010-027-FMS-05-00, Rev. 00Instructions for Installation ASR-2010-027-IF-10-00, Rev. 00Instructions for Continued Airworthiness ASR-2010-027-ICA-11-00(r&lt;n&gt;)Rev. 00 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>04/10/2011</td>
<td>Active</td>
<td>10030537</td>
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<td>10036590</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC ST00695AT-D</td>
<td>B747-400</td>
<td>Modification of Upper Deck Crew Rest Compartment in accordance with Timco Master Data List, MDL-01090, Revision E, dated July 22, 1996, or FAA approved Revision/Validation of FAA STC ST00695AT-D</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23 Amendment 2The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td></td>
<td>None</td>
<td>21/09/2011</td>
<td>Active</td>
<td>10030538</td>
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<tr>
<td>10036603</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC SA05-60</td>
<td>BD-700-1A11</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9427</td>
<td>EASA AM-809(2009)The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS-23 Amendment 2The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>TCCA STC No. SA05-60 Issue 1, dated 14 June 2005 Aerial Consulting Services Ltd Configuration Definition List E/25000Rev. FE, dated 20 September 2011 Bombardier Aerospace Aerial Plane Flight Manual Supplement G-FA/25000Rev. 01, dated 14 June 2005 Aerial Consulting Services Ltd Configuration</td>
<td>Restricted to ASN 9427</td>
<td>22/09/2011</td>
<td>Active</td>
<td>10030600</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10036604</td>
<td>MOTORFLUG BADERN BADEN GmbH</td>
<td>REV. 1</td>
<td>228-212</td>
<td></td>
<td>Installation of Mission Equipment for Maritime Surveillance.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection and/or a change to the associated noise and/or emission levels.</td>
<td>28145F103137-0601, Rev. 2_, AIRM Supplement 28145F103137-0701, Rev. 2_, Instruction for Continued Airworthiness 28145F103137-0702, Rev. 1_, TLMC Supplement 28145F103137-1001, Rev. 0_, Installation Manual</td>
<td>None.</td>
<td>11/10/2011</td>
<td>Active</td>
<td>10036604</td>
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<tr>
<td>10036609</td>
<td>ST ENGINEERING AEROSPACE LTD.</td>
<td>A320-200 SERIES</td>
<td>Ulal Airlines' A320 Cabin Reconfiguration</td>
<td></td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. JAR 25 Change 11 and EASA CS 25 Amendments 4 and 10, as per ST Aerospace's Certification Plan Doc 399001/1 AED/001 Revision D. In addition, EASA REP 28145A001/1, Rev. 0, Installation Manual later revisions of the above listed documents approved by EASA.</td>
<td>ST Aerospace's Master Document List 399001/1 AED/002 Revision B, dated 22nd September 2011, or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>22/09/2011</td>
<td>Active</td>
<td>10036609</td>
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<tr>
<td>10036624</td>
<td>TECH-TOOL PLASTICS, Inc.</td>
<td>FAA STC SR03641AT</td>
<td>EC 120 B</td>
<td></td>
<td>Replace the existing Eurocopter EC120 B windows with Tech-Tool Plastics, Inc. replacement windows</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List (MDL) EC120/130-2000-100, Revision C dated October 31, 2009, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). Eurocopter EC 130-B Maintenance and Repair Manual, is also required.</td>
<td>None.</td>
<td>23/08/2011</td>
<td>Active</td>
<td>10036624</td>
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<tr>
<td>10036630</td>
<td>NSE INTEGRATIONS</td>
<td>ATR 42-300</td>
<td>Modification in ATR42-300 MSN 049's calibration test bench</td>
<td></td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. In addition to the product's original certification basis, EASA Certification Review Item (CRI) Special Condition CRI H-01 Issue 1 &quot;Enhance...</td>
<td>NSE Inauguration's Modification Form NSE 08-01 Edition 3, Revision 1 dated 20th September 2011 and Supplement for EWIS Instructions for Continued Airworthiness: &quot;Detailed Inspection Tasks ATR42 Kit Calibration Test Bins&quot; revision Indice B dated 20th September 2011 or later revisions of the above listed.</td>
<td>None.</td>
<td>28/09/2011</td>
<td>Active</td>
<td>10036630</td>
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<td>10036633</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>REV. 1</td>
<td>BD-700-1A11 BD-700-1A10</td>
<td></td>
<td>Dual Electronic Flight Bag (EFB) Installation</td>
<td>EASA IM.A.009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>RUAG Master Document List No. 341A-MDL Rev. N/C dated 15 September 2011, RUAG Approval for Aircraft Modification No. 341A-CN Rev. N/C dated 15 September 2011, RUAG Airplane Maintenance Manual Supplement No. 341A-AMMS or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>27/09/2011</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>CAU UN RESPONSIBIL ITY</td>
<td>CL-600-2B16</td>
<td>Introduction of Corporate Interior</td>
<td>EASA MA-023The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ap...</td>
<td>STC Twenty One Master Document List No. 921.MD-0064 Initial issue, dated 22 September 2011</td>
<td>Restricted to ASN 5616</td>
<td>26/02/2011</td>
<td>Active</td>
<td>10036644</td>
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<td>HAE.CO CABIN SOLUTIONS, LLC</td>
<td>FAA STC</td>
<td>ST01958AT-D</td>
<td>Installation of a Ball Camera</td>
<td>The Certification Basis for the original product remains applicable in this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>FAA STC ST01958AT-D</td>
<td>none</td>
<td>27/09/2011</td>
<td>Active</td>
<td>10036655</td>
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<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>ST02876CH</td>
<td>Installation of Vacuum Toilet Assembly</td>
<td>AMM Supplement 38-30-06 Rev.06 Instructions for Continued Airworthiness (ICA) and E915 ICA source document, dated 20 October 2014</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/06/2016</td>
<td>Active</td>
<td>10036659</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>SD3-SHERPA, SD3-60</td>
<td>GARMIN GPS-400</td>
<td>Installation of GARMIN GPS-400</td>
<td>CS 25 Amendment 4The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval</td>
<td>Scandinavian Avionic/Engineering Order 2042795-EO Issue 1 dated 21 July 2011</td>
<td>None</td>
<td>28/09/2011</td>
<td>Active</td>
<td>10036673</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>60, A60, B60</td>
<td>Installation of Traffic Advisory System</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable in this certificate/approval CS-23, Amendment 2The requirements for environmental protection and the associated/certified noise and/or emissions levels of the orig...</td>
<td>Flight Manual Supplement ASR-2011-016-FMS-05-00, Rev. 00-Instructions for Installation ASR-2011-016- IF1-10-00, Rev. 00-Instructions for Continued Airworthiness ASR-2011-016-ICA-11-00, Rev.00 for later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>28/06/2011</td>
<td>Active</td>
<td>10036677</td>
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<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td>PA-46R-350T</td>
<td>Installation of Tactical Weather Detection System (Stormscope)</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable in this certificate/approval CS-23, Amendment 2The requirements for environmental protection and the associated/certified noise and/or emissions levels of the orig...</td>
<td>Flight Manual Supplement TWX670- NY-02180-08, Rev. 1</td>
<td>This STC is also applicable to Turbocharged Aircraft (PA-46-310P,PA46-350DLX) according to FAA STC ST00541SE. The aircraft must have been already equipped with a suitable MultiFunction Display (MFD). Prior to installation of this modification it must be determined that the interrelationship betw...</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10036680</td>
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<td>STC Holder</td>
<td>Rev.</td>
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<td>Models</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>SOCATA SAS</td>
<td>REV. 1</td>
<td>AS 355 F2</td>
<td></td>
<td>Installation of a surveillance WESCAM MX 10 camera and operator console...</td>
<td></td>
<td>MANUFACTURER CERTIFICATION OF...</td>
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<td>17/12/2014</td>
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<td>TATENHILL AVIATION LTD</td>
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<td>Design Data for the approval of a Honeywell KFD840 Primary Flight Display...</td>
<td>CS-23 Amendment 1</td>
<td>Installation Manual, dated 09...</td>
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<td>04/10/2011</td>
<td>Active</td>
<td>10036731</td>
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<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>REV. 1</td>
<td>FAA STC SA106650C</td>
<td>B300, B300C</td>
<td>Installation of upgraded hardware to the Rockwell Collins Pro Line 21 Flight...</td>
<td></td>
<td>Master Drawing List, dated 12...</td>
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<td>04/12/2020</td>
<td>Active</td>
<td>10036735</td>
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<td>BEECHCRAFT CORPORATION</td>
<td></td>
<td>FAA STC SA0231/1A</td>
<td>B300 &amp; B300C</td>
<td>Installation and maintenance of Switches and wiring for an Electro-Optical/Infra-Red Lift and Tunnel System.</td>
<td>FAR 23 The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Installation in accordance with Master Data List 3000E049584, Rev. 3, dated November...</td>
<td></td>
<td>05/10/2011</td>
<td>Active</td>
<td>10036742</td>
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<td>NORTHERN AVIONICS S.R.L.</td>
<td>REV. 1</td>
<td>A109A, A109 All, A109C</td>
<td></td>
<td>Installation of a retractable swivelling landing light (searchlight).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/04/2013</td>
<td>Active</td>
<td>10036765</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>FAA STC/ SA00857WI</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>10036781</td>
<td>HAWKER BEECHCRAFT SERVICES</td>
<td>BEECH 1900D</td>
<td>Installation of Universal Avionics System Corp UNS-1K Flight Management System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate/ approval remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Raytheon Aircraft Services MasterDrawing List ref. 148-100013, Revision B, dated February 24, 2000. Operation in accordance with Raytheon Aircraft Services Flight Manual Supplement ref. FAAST0249UE388, dated March 2, 2000 or later revisions of the above listed documents.</td>
<td>None.</td>
<td>06/10/2011</td>
<td>Active</td>
<td>10036781</td>
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<tr>
<td>10036782</td>
<td>HAWKER BEECHCRAFT SERVICES</td>
<td>BEECH 1900D</td>
<td>Installation of a forward double or triple club executive style interior configuration.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate/ approval remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Raytheon Aircraft Services MasterDrawing List ref. RAS19D1000, Revision A, dated May 24, 2000. Operation in accordance with Raytheon Aircraft Services Flight Manual Supplement ref. RAS19001FMS, dated May 24, 2000 or later revisions of the above listed documents.</td>
<td>None.</td>
<td>06/10/2011</td>
<td>Active</td>
<td>10036782</td>
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<tr>
<td>10036784</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>BEECH 60, A60, AND B60</td>
<td>Installation of Garmin G600 System with Synthetic Vision Technology (SVT).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate/ approval remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Avionik Straubing Engineering Order refASR-2011-015, Amendment 00 dated 4/4/11. Operation in accordance with Avionik Straubing Flight Manual Supplement ref. ASR-2010-015-FMS-05-00, Amendment 00, dated 4/4/11. Instructions for Continued Airworthiness in accordance with Avionik Straubing Flight Manual.</td>
<td>None.</td>
<td>06/10/2011</td>
<td>Active</td>
<td>10036784</td>
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<tr>
<td>10036788</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td>REV. 1</td>
<td>Installation of a partition in the rear cabin area.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate/ approval remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with SEGA Modification File A-25-Cloison_DA dated September 2011. Instructions for continued airworthiness as defined in SEGA documentation A-25-Cloison &lt;(n+)&gt; dated September 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>03/01/2012</td>
<td>Active</td>
<td>10036789</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>H-01 EWIST</td>
<td>INNOTECH AVIATION</td>
<td>REV. 3</td>
<td>TCCA STC NOT0213</td>
<td>MDL-BD700-148</td>
<td>The Certification Basis (CB) for the original product remains applicable. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EASA STC no. 10036806</td>
<td>LDA01/11/L200</td>
<td>None</td>
<td>27/01/2016</td>
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<td>10036806</td>
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<tr>
<td>10036826</td>
<td>DSA A.S.</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td></td>
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<td></td>
<td>08/05/2020</td>
<td>Active</td>
<td>10036826</td>
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<tr>
<td>10036814</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 5</td>
<td>CAA UK CAA, 21ME2.00057</td>
<td>757-200</td>
<td>FAR Part 21 Validation for EASA STC CAA.21ME2.00057 Rev 4. Transfer from FAA STC SR05215E SEE EASA APPROVED MODEL LIST Installation of Geneva Aviation Audio System on CS-27/Part-27 Category Rotorcraft in accordance with the enclosed EASA Approved Model List (AML). The purpose of this revision 2 is to move CS-29/Part 29 Category Rotorcraft from this approval. Please see enclosed (page 5 of 7 of this document) EASA approved CS-27/Part 27 Rotorcraft model list (AML) for Geneva Aviation Audio System/ROTORCRAFT: FLIGHT MANUALS SEE EASA APPROVED MODEL LIST or later revisions of the above listed documents approved by EASA.</td>
<td></td>
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<td>28/04/2021</td>
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<td>10036823</td>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD</td>
<td>REV. 3</td>
<td>TCCA STC NO. SAI0-8</td>
<td>8D700-1A10-1A11 BD700-1A10-1A11</td>
<td>The Certification Basis (CB) for the original product remains applicable. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>13/03/2020</td>
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<td>10036823</td>
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<td>10036806</td>
<td>GENEVA AVIATION Inc.</td>
<td>REV. 2</td>
<td>FAA STC SR05215E</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Geneva Aviation Audio System on CS-27/Part-27 Category Rotorcraft in accordance with the enclosed EASA Approved Model List (AML). The purpose of this revision 2 is to move CS-29/Part 29 Category Rotorcraft from this approval. Previously included CS-29/Part 29 Category Rotorcraft are re...</td>
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<td></td>
<td>27/01/2016</td>
<td>Active</td>
<td>10036806</td>
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<td>10036793</td>
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<td>FAA STC Master Data List No. AES-757-172MDL, Issue 1, dated 20 September 2011 or later revisions of the above listed documents approved by EASA.</td>
<td></td>
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<td>None</td>
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<td>STC Holder</td>
<td>STC No</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>POLICE AVIATION SERVICES Ltd.</td>
<td>10036834</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>VULCANAR P 68 TC OBSERVER</td>
<td>Airborne Surveillance System Installation</td>
<td>CS-23 Amendment 2 dated 28 September 2010The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable...</td>
<td>Modifications to the Certificate of Design PAS/ MCD/M43/Compliance Checklist CCL/943-003 Issue 8Maintenance Schedule Supplement MSS/943-007 Issue Aircraft Flight Manual NDR.10.707-4B Supplement FMS/943-002 Issue 8or later revisions of the above listed documents approved by EASA None</td>
<td>24/11/2011</td>
<td>Active</td>
<td>10036834</td>
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<td>BE Aerospace, Inc.</td>
<td>10036838</td>
<td>FAA STC ST03918AT</td>
<td>777-300/300ER 777-200/200LR</td>
<td>FAA STC ST03918AT: Installation of the Honeywell MCS-7200 SATCOM System Revision 2 transferred from EMTEQ Engineering to BE Aerospace. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23, Amendment 2The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable...</td>
<td>FAA STC ST03918AT: MDL: 78144-001 Rev. A dated 13 May 2011-CA 2-78144/804 Rev: IR dated 10 Feb 2011for later revisions of the above listed documents approved by EASA Pre to instaltion of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10036838</td>
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<tr>
<td>AVIONIK STRAUBING ENWORKINGS GmbH</td>
<td>10036858</td>
<td></td>
<td>60, A60, B60</td>
<td>Installation of a Satellite Data Link System Garmin GSP50i. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CS 23, Amendment 2 The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable...</td>
<td>Flight Manual Supplement: ASR-2011-051-FMS-05-00, Rev. 00Instructions for Installation: ASR-2011-051-IR-10-00, Rev. 00Instructions for Continued Airworthiness: ASR-2011-051-KCA-11-00, Rev. 00for later revisions of the above listed documents approved by EASA None.</td>
<td>12/10/2011</td>
<td>Active</td>
<td>10036858</td>
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<tr>
<td>CHRISTEN- AIRTECH AG</td>
<td>10036867</td>
<td>FAA STC SR064A3SE</td>
<td></td>
<td>Inst. Traffic alerting system/ Stormscope Removed: - NIL Installed - Garmin GT580 Traffic Alerting System- L-3 WX500 Stormscope System CS 23, 23: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CS 23, Amendment 2 The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable...</td>
<td>Master Document List: Doc. No. CAT-211-2694- EOMDL-01. Rev. 00Instructions for continued airworthiness: For the Garmin GT580, Doc. Nr. CAT-211-2694-CA-01 For the L-3 WX500, Doc. Nr. CAT-211-2694-CA-02 Rev. 0 None.</td>
<td>13/10/2011</td>
<td>Active</td>
<td>10036867</td>
<td></td>
<td></td>
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<tr>
<td>Eagle Copters USA, Inc.</td>
<td>10036888</td>
<td>FAA STC SR064A3SE</td>
<td></td>
<td>Installation of Geneva Aviation Video System per MDL GA183 Notes 1. The STC was initially issued to Geneva Aviation, Inc. (installation reference: 10036888). 2. Revision 1 to this STC reflects the transfer of 10036868 from Geneva Aviation, Inc. to Eagle Copters USA, Inc. The Certification Basis (CS) for the original product remains applicable to this certificate/approval: The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List GA183 Rev. K dated May 22, 2010Rotorcraft Flight Manual Supplement for AAS350/355/360 (C) and (B) GA183-3 Rev. Dated December 10, 2009 (for AS350/AS555 model aircraft)Rotorcraft Flight Manual Supplement for BO-105 (C) and (B) GA183 Rev. Dated June 30, 2009 (for BO-105 model aircraft)Rotor... Any optional modifications listed in Installation Instructions referenced in Master Drawing List GA183, to be installed in conjunction with this STC, shall be EASA approved. This STC covers installation of the complete set of equipment items as defined by this STC or a subset of these equipment items... Any optional modifications listed in Installation Instructions referenced in Master Drawing List GA183, to be installed in conjunction with this STC, shall be EASA approved. This STC covers installation of the complete set of equipment items as defined by this STC or a subset of these equipment items... None.</td>
<td>31/03/2015</td>
<td>Active</td>
<td>10036888</td>
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<td></td>
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<tr>
<td>AEROLITE AG</td>
<td>10036912</td>
<td>Airbus A340/343</td>
<td>AEW/3913/BS139</td>
<td>EMS Equipment PN: 139-003870-501 The STC covers 4 EMS cabin configurations and 3 PAX configurations as described in the Masterdocument list MDL-11-016 rev N/C Rev. 1 changes the name of the STC holder from &quot;Aerolite Max Bucher AG&quot; to &quot;Aerolite AG&quot; The certification basis (CB) for the original product remains applicable to this certificate/approval: The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FMS: FMS-139-003870-501 rev N/C Airworthiness limitations WVR-139-003870-501-0 initial Issuer later revisions of the above listed documents approved by EASA/Operations manual: OPM-139-003870-501 rev A Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product None.</td>
<td>22/02/2019</td>
<td>Active</td>
<td>10036912</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10036936</td>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td>F27 MK60</td>
<td>Installation of cabin divider</td>
<td>The Certification Basis for the original product remains applicable to the associated certificate/approval. The requirements for environmental protection and the associated certificate/approval are included in the maintenance instructions according to EO 5117-07.</td>
<td>Prior to installation of this design change, it must be determined if the interrelationship between this design change and any other previously installed design changes or repair will introduce a risk effect upon the aircrafts' offshore product.</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10036936</td>
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<td>10036940</td>
<td>KUERZI AVIONICS AG</td>
<td>REV. 1</td>
<td>EC 130 T2</td>
<td>Installation of Honeywell TCAS II ATC</td>
<td>The Certification Basis for the original product remains applicable to the associated certificate/approval.</td>
<td>Master Document List 26/11/2011 REV B, dated 23 November 2018</td>
<td>At least a GPS/WAAS navigator, compatible with GS-100H, must be installed pre-requisite.</td>
<td>26/11/2018</td>
<td>Active</td>
<td>10036940</td>
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<td>10036941</td>
<td>TELAIR INTERNATIONAL, LLC</td>
<td>REV. 1</td>
<td>FAA STC ST02367LA</td>
<td>Installation of cabin divider</td>
<td>The Certification Basis for the original product and following additional or alternative airworthiness requirements are applicable to the associated certificate/approval.</td>
<td>PSCP 192100CPP-705, FAA STC 14220-A T and Index List 192100-8 REV B or later revisions of the above listed documents approved by EASA in accordance with FAA STC 14220-A T and Index List 192100-8 Revision B or later revisions of the above listed documents approved by EASA in accordance with FAA STC 14220-A T and Index List 192100-8.</td>
<td>None</td>
<td>20/10/2011</td>
<td>Active</td>
<td>10036941</td>
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<tr>
<td>10036942</td>
<td>TELAIR INTERNATIONAL, LLC</td>
<td>REV. 1</td>
<td>FAA STC ST02368LA</td>
<td>Installation of cabin divider</td>
<td>The Certification Basis for the original product and following additional or alternative airworthiness requirements are applicable to the associated certificate/approval.</td>
<td>PSCP 192100CPP-705, FAA STC 14220-A T and Index List 192100-8 REV B or later revisions of the above listed documents approved by EASA in accordance with FAA STC 14220-A T and Index List 192100-8.</td>
<td>None</td>
<td>20/10/2011</td>
<td>Active</td>
<td>10036942</td>
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<td>10036958</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>BAE 146, AVRO 146-RJ</td>
<td>Provision for a Dual NAVERO I Bag to Class 2 Electronic Flight Bag System</td>
<td>The Certification Basis for the original product and following additional or alternative airworthiness requirements are applicable to the associated certificate/approval.</td>
<td>SCANDINAVIAN Avionics Engineering Order and Certification Program: Provision for a Dual NAVERO I Bag to Class 2 Electronic Flight Bag System in a BAE 146 / Avro 146-RJ Series.</td>
<td>None</td>
<td>19/10/2011</td>
<td>Active</td>
<td>10036958</td>
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<tr>
<td>10036974</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 1</td>
<td>C90A</td>
<td>Installation of Honeywell TCAS II ATC and GPMIS MK VI according to to manufactured modification n° MOD 1085.</td>
<td>The Certification Basis for the original product and following additional or alternative airworthiness requirements are applicable to the associated certificate/approval.</td>
<td>Flight Manual Supplement n° AE1085SMV Revision 1 dated 21/03/2011 for continued airworthiness n° AE1085SMW revision 1 dated 11/04/2011.</td>
<td>None</td>
<td>21/10/2011</td>
<td>Active</td>
<td>10036974</td>
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<td>STC no.</td>
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<td>Rev. level</td>
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<td>Movies</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of issue</td>
<td>Status</td>
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<td>10036985</td>
<td>CESSNA ZURICH CITATION SERVICE CENTGBBH</td>
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<td></td>
<td>Removed Equipment: Aerosonic VSI 1 &amp; 2-Skylight 497 TCAS Systems; new installed equipment: Honeywell TAWS Indicator 1 &amp; 2; PS-578A XPOS/TCAS Control Head; CAS-67A TCAS II System</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. SCs or ESFs; Voluntary compliance to CS-23 First Issue for the changes/Derogation to latest amendment (Part 21A.101). Do not materially change.</td>
<td>Master Document List Doc. No. MDL-075, Rev. B-ICA, Doc.No: ICA-040, Rev. A; AFMS EHS, Doc. No. FMS-042, Rev. 9(OR later revisions of the above listed documents approved by EASA).</td>
<td>S/N: 525-0322 F-HCP6</td>
<td>24/10/2011</td>
<td>Active</td>
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<td>10037000</td>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td>EASA.A.S.03714</td>
<td></td>
<td></td>
<td>Installation of Cabin Interior Doors in accordance with JetAviationMaster Data List BHHM1001-001 Rev. B, dated 19th October, 2011</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>BHHM1001-201-02 Rev. IR - Airplane Flight Manual Supplement BHHM1001-15-02 Rev. IR - Instructions for Continued Airworthiness</td>
<td>None.</td>
<td>24/10/2011</td>
<td>Active</td>
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<tr>
<td>10037001</td>
<td>AERO TWIN, Inc.</td>
<td></td>
<td>FAA STC</td>
<td>SA4626NM</td>
<td></td>
<td>Fabrication and Installation of Auxiliary dry air pump kit AP8-100 (inaccordance to FAA STC SA4626NM.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement, Rev. B, dated November 27, 2009Instructions for Continued Airworthiness, Doc. No. AP8-ICA, Rev. D, dated June 22, 2009Master Drawing List No. AP8-100, dated October 13, 2009Installation Instructions, Doc. No. AP8-149, Rev. F, dated October 12, 2009 or later revisions of.</td>
<td>None.</td>
<td>24/10/2011</td>
<td>Active</td>
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<tr>
<td>10037002</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING RPORATION AG (AEC)</td>
<td></td>
<td>CESSNA 208, 208B</td>
<td></td>
<td></td>
<td>Cabin Interior Removal and Modification.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Modification in accordance with Master Data List Doc. No. 5143-049, Rev. Instructions for Continued Airworthiness</td>
<td>None.</td>
<td>24/10/2011</td>
<td>Active</td>
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<tr>
<td>10037011</td>
<td>RUAG AG</td>
<td>REV. 1</td>
<td>EC 130 B4</td>
<td></td>
<td></td>
<td>Avionics RetrofitOut: Garmin GTX 330 Mode S Transponder &amp; GMA 340 Audio Marker system/Honeywell XX 106A-VHF(COMNAV) system; Asapn EFD 1000H EFS10 system; Garmin GPS 330 VHF COM/NAV &amp; GPS system; Garmin GSR 56 Indum system; PS Engineering PMA 8000Audio/Marker system; Avionics TAD 620 system; Beech...</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Flight Manual Supplement, FMS-1900819-01 Rev. AMaster Document List: MDL-1900815-01 Rev. Instructions for Continued Airworthiness ICA-1900815-01 Rev.</td>
<td>S/N: 7214 Prior to installation of this change/repair it must be determined if the relationship between this change/repair and any otherwise installed design change/repair and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/02/2021</td>
<td>Active</td>
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<tr>
<td>10037015</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td>A319-133</td>
<td></td>
<td></td>
<td>Airbus Corporate Jet Centre Modification C/JO700 Issue 3-WP Interior - OPTIMA PROJECT/Revision due to update of AFM-Sand MPP-5</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements as listed in the General CRI A-4301-004 NGIC Project Certification Basis. The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>-, Modification Approval Sheet, ref. C/JO700, Issue 3, dated 24th October 2013...Airplane Flight Manual - AFM - Supplement, ref. FM-S MOD C/JO700, Revision 02, dated 1st October 2013...Maintenance Planning Document - MPD - Supplement, ref. MPD-S MODC/JO700, Revision 02, dated 1st October 2013...Master M...</td>
<td>This STC is limited to A319-133 MSN 4319 Prior to installation of this design change it must be determined if the relationship between this design change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/10/2013</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC no</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>B/E AEROSPACE, Inc.</td>
<td>REV.1</td>
<td>FAA STC ST03253AT</td>
<td>757-200F, 757-300 757-200, 757-200CB</td>
<td>Installation of the Honeywell MK II Aircraft Communication Addressing and Reporting System (ACARS) Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement 2-35716-807 Revision IR, dated 20 April 2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/004 CF (or subsequent revisions of this decision). EMTEQ Engineering Master Drawing List 2-35716-001 Revision C, dated 19November.</td>
<td>Compliance with EASA AMC 20-0-9 and AMC 20-10 has not been shown Prior to installation of this design change it must be determined whether the relationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness...</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10037018</td>
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<tr>
<td>AEROTEC &amp; A CONCEPT</td>
<td>REV.1</td>
<td>C90A</td>
<td></td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval CS 23 amdt 1: 301, 303, 305, 337, 361, 571, 603, 607, 771, 773, 867, 1301, 1303, 1309, 1311, 1321, 1322, 1323, 1325, 1327, 1351, 1353, 1359...</td>
<td>Flight Manual Supplement No. AE10848MV revision 2 dated 21/10/2011 Flight Manual Supplement No. AE1084155MV revision 0 dated 03/04/2013 Instructions for Continued Airworthiness No. AE1084PRM revision 1 dated 11/04/2011 Master Data List No. AE10846MDL revision 4 dated 21/10/2011 Master Data List No. AE12524.</td>
<td>1. This change is applicable to aircraft with Li 1066 and its approved IFR RNAV, RNAV approach LNAV and LPV operations. 2. RNP approach with APV bare VNAV is not authorized. 3. The airplane must be equipped with - an altitude indicator, an aisapped indicator and an altimeter independent from the original...</td>
<td>16/05/2014</td>
<td>Active</td>
<td>10037027</td>
<td></td>
<td></td>
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<tr>
<td>SASU S AIR ENTREPRISE</td>
<td></td>
<td>200</td>
<td></td>
<td>Installation of gear override and steep approach switches on TWASTW-950.</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The certificate/approval is not authorized. 3. The airplane must be equipped with - an altitude indicator, an aissapped indicator and an altimeter independent from the original...</td>
<td>Flight Manual Supplement No. UAE-ATFSM-11-002 revision IR dated 18/07/2011 Master Data List No. MCL-09-11 revision 1 dated 13/01/2021 Service Bulletin No. SB-UAE-005-11-01 dated 18/07/2011 or later revisions of the above listed documents approved by EASA.</td>
<td>This STC is applicable to aircraft with BY-91.</td>
<td>26/10/2011</td>
<td>Active</td>
<td>10037035</td>
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<tr>
<td>BOMBERGARDIER INC.</td>
<td></td>
<td>BD-700-1A10</td>
<td></td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9413</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The certificate/approval is not authorized. 3. The airplane must be equipped with - an altitude indicator, an aissapped indicator and an altimeter independent from the original...</td>
<td>TCCA STC No. SA44-113 Issue 1, dated 28 October 2004.</td>
<td>Restricted to ASN 9413 and 1. Refer to the Limitations sections of the required AFM Supplement and/or the relevant FCOMs. 2. VNAV System is compatible only to IEEE 802.11.b.</td>
<td>26/10/2011</td>
<td>Active</td>
<td>10037040</td>
<td></td>
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<tr>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td></td>
<td>FAA STC ST096406E</td>
<td>B747-400</td>
<td>Recomfiguration of the Main Deck of the B747-400 series airplane inaccordance with Flight Structures, Inc. Master Drawing List 9F699028-2001 Rev. E, dated 10 November 2009 or a later FAA approved revision Validation of FAA STC ST096406E ...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Flight Structures, Inc. Master Data List MCL 9F699028-2001 Rev. E or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/004 CF (or subsequent revisions of this decision).</td>
<td>None.</td>
<td>27/10/2011</td>
<td>Active</td>
<td>10037049</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NASA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>BOMBARDIER INC.</td>
<td></td>
<td></td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9238</td>
<td>EASA &amp; A.009 The Certification Basis for the original product remains applicable both certificate/approval The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TC-CA STC No. SA04-113 Issue 1, dated 28 October 2004. Aeronautical Services Configuration Definition List DB54000 Revision J1, dated 07 October 2011. Instructions for Continued Amorphous Device GC3923000001-FC0 or later revisions of the above listed documents approved by EASA in accordance with EASA ED D..</td>
<td>Restricted to ASN 9238 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced COMIS 2. WLAN System is compatible only to IEEE 802.11(b).</td>
<td>27/10/2011</td>
<td>Active</td>
<td>10037055</td>
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<tr>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td></td>
<td>F900B, F900C &amp; F900EX</td>
<td>Cabin Master installation. A Cabin Master Switch is installed on Copilot console.</td>
<td>The Certification Basis for the original product remains applicable both certificate/approval The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>DFS Change Approval Sheet DF5648-CAS Issue A dated 30.06.2011 DFS Modification Sheet DF56964, Issue A dated 30.06.2011 DFS Airplane Flight Manual Supplement N° FS01-01 dated 30.06.2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED D..</td>
<td>Not applicable</td>
<td>27/10/2011</td>
<td>Active</td>
<td>10037056</td>
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<tr>
<td>TECH-TOOL PLASTICS, Inc.</td>
<td></td>
<td></td>
<td>EC130 B4</td>
<td>Installation of Replacement Windows This STC is the validation of FAA STC SR03940AT</td>
<td>The Certification Basis for the original product remains applicable both certificate/approval The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Tech-Tool Plastics, Inc. Master Drawing List (MDL), EC120/130-2000-100 Revision C dated October 31, 2000 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/14/OF (or subsequent revisions thereof decision).</td>
<td>None.</td>
<td>31/10/2011</td>
<td>Active</td>
<td>10037088</td>
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<tr>
<td>ROCHWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>B747-400</td>
<td>Installation of Collins SAT-906 Satcom Systems provisions in accordance with Timco Master Data List MDL SR03940AT Issue C, dated October 21, 1999 or later approved FAA revisions.</td>
<td>The Certification Basis for the original product remains applicable both certificate/approval The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rockwell Collins RCA-280 Rev. - dated March 5, 2010 Instructions for continued airworthiness Damage Tolerance Inspection Timco Maintenance Manual Supplement to Chapter 25-00-00-001 September 15, 1999 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/14/OF (or subsequent revisions thereof decision).</td>
<td>None.</td>
<td>13/04/2012</td>
<td>Active</td>
<td>10037095</td>
<td></td>
</tr>
<tr>
<td>BOMBARDIER INC.</td>
<td></td>
<td></td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9420.</td>
<td>The Certification Basis for the original product remains applicable both certificate/approval The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TC-CA STC No. SA04-113 Issue 1, dated 28 October 2004. Aeronautical Services Configuration Definition List DB54000 Revision J1, dated 07 October 2011. Instructions for Continued Amorphous Device GC3923000001-FC0 or later revisions of the above listed documents approved by EASA in accordance with EASA ED D..</td>
<td>Restricted to ASN 9420 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced COMIS 2. WLAN System is compatible only to IEEE 802.11(b).</td>
<td>02/11/2011</td>
<td>Active</td>
<td>10037096</td>
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<tr>
<td>GULFSTREAM AEROSPACE</td>
<td></td>
<td>FAA STC</td>
<td>G-IV-SP</td>
<td>Executive Aircraft Completion, Multiple STC.</td>
<td>The Certification Basis for the original product remains applicable both certificate/approval The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List GC5220040000, Revision 1, dated October 18, 2011. Operation to be in accordance with Gulfstream Flight Manual Supplement GC5220040000 Original Release, dated October 18, 2011. The following Instructions for Continued Airworthiness...</td>
<td>None.</td>
<td>03/11/2011</td>
<td>Active</td>
<td>10037106</td>
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**Note:** The information provided is a screenshot of a document listing EASA Supplemental Type Certificates. The details include STC holder, model, description, certification basis, associated technical documentation, limitations/conditions, date of issue, status, and system record.
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<tr>
<th>STC Holder</th>
<th>Rev. Level</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>BOMBARDIER AEROSPACE</td>
<td>REV. 1</td>
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<td>TCCA STC NO. C-LSA05-030/D</td>
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<td>Transport Canada STC No. C-LSA05-030/D Issue 5 dated 01 November 2011</td>
<td>Airworthiness ref. CRI-D FN2500121 Revision NC or later approved</td>
<td>04/11/2011</td>
<td>Active</td>
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<tr>
<td>ROEIDER INCA CORPORATION</td>
<td>REV. 1</td>
<td>Master Document List BK02-4420/Engineering Order EG-BK-4420 or later revisions of the above listed documents approved by EASA</td>
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<td>04/11/2011</td>
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<td>DELTA ENGINEERING</td>
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<td>CORPORATION</td>
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<td>STC No.</td>
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<td>Associated Technical Documentation</td>
<td>Date of Issue</td>
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<tr>
<td>10037167</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>Installation of Air Ambulance Equipment - Extension of existing EASA STC No. SA04-113</td>
<td>The Certification Basis for the original product remains applicable to the Class TBC stated in the STC. The requirements for environmental protection and the associated certificated noise and/or emissions remain applicable to this certificate/approval.</td>
<td>Master Data List Rev. 0, MDL-Bell 412-1, CR-05.08.2011, Compliance Record Rev. 0, CR-MF50-17-000-5xx, Date: 06.08.2011</td>
<td>10/06/2011</td>
<td>Active</td>
</tr>
<tr>
<td>10037201</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>Installation of Air Ambulance Equipment - Extension of existing EASA STC No. SA04-113</td>
<td>The Certification Basis for the original product remains applicable to the Class TBC stated in the STC. The requirements for environmental protection and the associated certificated noise and/or emissions remain applicable to this certificate/approval.</td>
<td>Master Data List Rev. 0, MDL-Bell 412-1, CR-05.08.2011, Compliance Record Rev. 0, CR-MF50-17-000-5xx, Date: 06.08.2011</td>
<td>10/06/2011</td>
<td>Active</td>
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<tr>
<td>10037233</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>Installation of Air Ambulance Equipment - &quot;quick conversion&quot; in MFS0 infra RH Cabin Section of Aircraft.</td>
<td>The Certification Basis for the original product remains applicable to the Class TBC stated in the STC. The requirements for environmental protection and the associated certificated noise and/or emissions remain applicable to this certificate/approval.</td>
<td>Master Data List Rev. 1, MDS-MF50-17-000-5xx, Date: 20.10.2011</td>
<td>10/11/2011</td>
<td>Active</td>
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<td>10037255</td>
<td>ANUVU</td>
<td>Implementation of Electronic Flight Bag Class 2 TBC</td>
<td>The Type and OSD Certification Bases for the original product remain applicable to the Class TBC stated in the STC. The requirements for environmental protection and the associated certificated noise and/or emissions remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>04/07/2022</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Type</td>
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<td>Limitations/Conditions</td>
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<td>10037296</td>
<td>HAWKER BEECHCRAFT LTD.</td>
<td>RE-1</td>
<td>CAAS RESPONSIBILITY</td>
<td>To introduce interior and electrical provisions for Medics stretcher/mobility change capability. This Change instructs the interior and electrical changes to the aircraft. Stretchers and medical equipment installation will be certified to the appropriate vendor.</td>
<td></td>
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<tr>
<td>10037297</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC SAUS-60</td>
<td>BD-700-1A11</td>
<td>Installation of Complete Custom Aircraft Interior - ASN 9430.</td>
<td>EASA M.4980 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. SAUS-60 Issue 1 dated June 2006. Aerocustom Services Ltd. Configuration Definition List. EO/25/00 Revision FJ dated 10 November 2011. Bombardier Aerospace Aerial Flight Manual Supplement G-FM/250045-01 Revision 2. or later approved revision. GCD/430/0000-IFCA Instructions for continued operation.</td>
</tr>
<tr>
<td>10037304</td>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td>PA-44-180</td>
<td>Avionics upgrade including Aspen Electronic Flight Display (EFD) 100/Pro system and GNS 430 W with coupling to S-TEC 55X automatic pilot</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CS 23 original issue, § 303, 305, 561, 601, 603, 607, 609, 867, 1301, 1501, 1311, 1312, 1329, 1351, 1353, 1357, 1365, 1381, 1431, 150.</td>
<td>SEAM Flight Manual Supplement SM/1 dated 08/12/2009 and SM/2 dated 20/05/2009. or later approved revision. The only authorized GPS mode for the S-TEC 55X automatic pilot (AP) is NAV mode.</td>
<td>17/11/2011</td>
</tr>
<tr>
<td>10037313</td>
<td>DAHER SOCATA A/S</td>
<td>AS355 F2</td>
<td>Modification STC 34.001.11 Version A for IFR - Installation of GNS 430W for VH...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FMS-STA 34.001.11 Version A IFR - Flight Manual Supplement. Rev. - FMS-STA 34.001.11 Version B IFR - Flight Manual Supplement. Rev. - or later revisions of the above listed documents approved by EASA-STC 34.001.11 Version A VFRB FRA - Installation Description Rev. 1-STC 34.001.11 Version A VFRB FRA</td>
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<td>10037327</td>
<td>GPI</td>
<td>AS 332 C, C1, L, L1, L2</td>
<td>Installation of Cargo Net in Cabin.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>SFM-P-100-07 issue 1 dated 3 November 2011 - Flight Manual Supplement.</td>
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<td>STC Holder</td>
<td>STC ref.</td>
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<td>Certification basis</td>
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<td>Limitations/Conditions</td>
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<tr>
<td>SCANIAVIAN AVIONICS DESIGN ApS</td>
<td>CESSNA 525B</td>
<td>Installation of L3 FA2100 Digital FDR</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>EASA.</td>
<td>SN 526B-159 and on</td>
<td>18/11/2011</td>
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<td>ATELIERS CHАBORD</td>
<td>FA150K, FA150L, F150K, F150L, F150M, F150G, F150H, F150J</td>
<td>STC CESSNA150-11-01 Rev. E - Dispositif de tolérance radioélectrique de la cabine</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAR 23, Amdt. 0:23.21, 23.147, 23.171, 23.179, 23.233, 23.251, 23.303, 23.305, 23.307, 23.321, 23.347, 23.365, 23.369, 23.609, 23.627, 23.639, 23.641, 23.661, 23.669, 23.679, 23.689.</td>
<td>22/12/2017</td>
<td>Active</td>
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<tr>
<td>CRANFIELD AEROSPACE LTD</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>CESSNA 421C</td>
<td>Installation of Initial Roll Change Equipment in a Civil Contingency Aircraft - Cessna 421C as defined in Cranfield Aerospace document CAaM/Cessna/1327.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval FAR 23, Amdt. 0:23.21, 23.147, 23.171, 23.179, 23.233, 23.251, 23.303, 23.305, 23.307, 23.321, 23.347, 23.365, 23.369, 23.609, 23.627, 23.639, 23.641, 23.661, 23.669, 23.679, 23.689.</td>
<td>Type Design Definition CAI/DD/0106AFMS: Supplement No. 2 to POH D1546-5-13; Instructions of Continued Airworthiness CAA/ML/1327.</td>
<td>22/11/2011</td>
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<td>CRANFIELD AEROSPACE LTD</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>CESSNA 421C</td>
<td>System and Structural Provisions for a Civil Contingency Aircraft - Cessna 421C as defined in Cranfield Aerospace document CAaM/Cessna/1325.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval FAR 23, Amdt. 0:23.21, 23.147, 23.171, 23.179, 23.233, 23.251, 23.303, 23.305, 23.307, 23.321, 23.347, 23.365, 23.369, 23.601, 23.603, 23.609, 23.627, 23.639, 23.641, 23.661, 23.669, 23.679, 23.689.</td>
<td>Type Design Definition CAI/DD/0106AFMS: Supplement No. 1 to POH D1546-5-13; Instructions of Continued Airworthiness CAA/ML/1326.</td>
<td>22/11/2011</td>
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<tr>
<td>TENECIA Ltd.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>A360-600 SERIES</td>
<td>Cockpit Door Monitoring System installation of the AD Aerospace &quot;CabinVu&quot; cockpit door monitoring system (video monitor, monitor control panel, video switch unit and three cameras) in accordance with Tenencia Ltd Modification Report/Master Data List) TEN 696 issue 3 and Certification Compliance Report...</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval ORI H-01 Special Condition for Enhanced Airworthiness Programme - Instructions for Continued Airworthiness on EWS...</td>
<td>Tenencia Maintenance Manual Supplement TEN 656/4 issue 1; Tenencia Operations Manual Supplement TEN 656/5 issue 1; Tenencia Installation Instructions (Configuration 1) TEN 656/10, issue 1; Tenencia Illustrated Parts Catalog TEN 656/17 issue 1; Tenencia Weight and Balance report TEN 656/24 issue 1 for later rev...</td>
<td>25/11/2012</td>
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<td>STC Holder</td>
<td>Certification basis</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>DASSAULT FALCON SERVICE</td>
<td>F900EX EASY F900B, F900C, F900EX</td>
<td>FR14 CABIN PARTITION WITH MECHANICAL POCKET SLIDING DOOR.</td>
<td>DFS Change Approval Sheet DF90664-CAS issue A dated 08.06.2011.</td>
<td>As per DFS Change Approval Sheet DF90664-CAS issue A dated 06.08.2011.</td>
<td>25/11/2011</td>
<td>Active</td>
</tr>
<tr>
<td>SIERRA INDUSTRIES Ltd.</td>
<td>FAA STC SA1435WE</td>
<td>Installation of dropped ailerons in flap down mode, re-contoured winglet edge in accordance with FAA STC SA1435WE.</td>
<td>Owner's Manual Supplement OMS 10-40Drawing List No. 10, dated June 25, 1985 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision). Maintenance instructions SMM 10-1.</td>
<td>SN 18051446 through 18052284 only. Installation of SC-100 Stall Warning system is not included.CD range: straight line between points: Amphibian with (+42.2) to (+43.9) @ 2950 lb.- EDG 5972790 (+36.0) to (+43.9) @ 2250 lb. or less. Control/Control Surfaces - Alternative: Paddle Setting.</td>
<td>29/11/2011</td>
<td>Active</td>
</tr>
<tr>
<td>INNOTECH AVIATION/ DIVISION OF IMP GROUP LTD.</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9416</td>
<td>EASA M.49/00 The Certification Basis for the original product remains applicable to this certificate/app.</td>
<td>Restricted to ASN 9416 Refer to the Airworthiness Limitations contained in the required Airplane Flight Manual Supplements. The Wireless LAN System is intended to provide internet connectivity and data services to the airplane's cabin using portable electronic devices (PEDs). Any other intended funds.</td>
<td>29/11/2011</td>
<td>Active</td>
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<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>TECH-TOOL PLASTICS, Inc.</td>
<td>FAAR STC SR09132RD-O</td>
<td>AS 355 N</td>
<td>Installation of a Vertical Sliding Window in the Right Handor Left Hand Sliding Door in accordance with Drawing List 350A40-1750, Revision B, dated 08/25/00.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>RFM 355N/1750 (B) dated March 19, 2001 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions thereof) and decision Drawing List 350A40-1750, Revision B, dated 08/25/00.</td>
<td>Both configurations cannot be installed concurrently (RightHand andLeft Hand)</td>
</tr>
<tr>
<td>PATS AIRCRAFT L.L.C.</td>
<td>REV. 1</td>
<td>FAA STC 701725NY-O</td>
<td>737-900ER</td>
<td>Installation of fan auxiliary fuel tank system configurations/approved configurations are the following: Configuration 1: Installation of a Five-Cell Auxiliary Fuel System in the forward and aft cargo compartments in accordance with PATS Aircraft, LLC. Kit Installation Drawing 8942-2811-100 (Kit No. -10...</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this approval: - CRH-1 issue 3 dated 30/11/2011. - CS 25.981.8(6) the requirements for environmental protection and the associated certificate noise and/or emission...</td>
<td>- Document TO-3688 revision IR dated 29 November 2011 Boeing 727-600ER/688(4) notes of applicable FARS and JARs (STC ST01725NY-O Validation) Certification plan and System Description TO-4688 revision A dated November 17, 2011. Certification plan and System Description TO-4750 revision B dated May 24...</td>
</tr>
<tr>
<td>JET AVIATION AG</td>
<td>REV. 5</td>
<td>FAA STC SR09132RD-O</td>
<td>737-700</td>
<td>Installation of an additional installation area of the EAS STC/AMT-3800 Satcom antenna on the forward fuselage between STA 480 and STAS05B crown area.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Plan: BSIL/M-00799ER-02 Issue 1 (&lt;-&gt; or later revisions of the above listed document(s) approved/accepted under the EASA system. Prior to installation of this change/repair it must be determined whether this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST03060AT</td>
<td>757-200-200CB/-200PFIN/-300</td>
<td>Installation of the Iridium Satcom Communication System in the Boeing 757-200 and 757-300 Series aircraft Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Drawing List 2-68511-011, Revision M, dated November 3, 2015. 2. FAA Approved Flight Manual Supplements for the Iridium Satcom Communications System are required parts of this STC. a. Configuration 1-2-64617-004, Revision A, dated November 15, 2011. b. Configuration 1-2-89132-839, Revision A...</td>
</tr>
<tr>
<td>WESTERN AVIONICS</td>
<td>DHC-6-1/-100/-200-300</td>
<td>Installation of Garmin GPS 155XLF for supplemental IFR Navigation in acc with TCCA STC SA01-93, Issue 1</td>
<td>Western Avionics Modification data Summary 71961-900, Revision 2, dated Mar 31, 2001. Western Avionics Flight Manual Supplement 71961-901, Issued Initial. TCCA Approved August 1, 2001 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subseq)...</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
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<td>10037522</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC NO. ST0177LA</td>
<td>CL-600-2B16</td>
<td>HUD/EVS Model 6605 Validation of HUD/EVS in a Deactivated State</td>
<td>EASA MA-0237The Certification Basis for the original product remains applicable to this certificate/app. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/app.</td>
<td>FAA STC NO. ST0177LA dated 04 December 2009 Installation of Rockwell Collins Model 6605 Head-Up Display (HUD) and Enhanced Vision System (EVS) CMA-2603 Infrared Sensor System (ISS) in accordance with section 9 of the Rockwell Collins Master Drawing list 9701-1267 Revision D-1.</td>
</tr>
<tr>
<td>10037524</td>
<td>SABENA TECHNICS BGC</td>
<td>A330-341/342/343, A330-351/352/353, A330-301/302/303</td>
<td>Panasonic eXPhone System activation on A330-300 with e21F(s) This STC installs a Panasonic eXPhone System allowing, in conjunction with other devices, the use of cellphones during flight.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/app.</td>
<td>Special Condition SC H-01 &quot;Enhanced Airworthiness Programme for Aeronautical Systems - ICA on EWIS The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/app.</td>
<td>Master Data List, Ref. 0496-02-A-2103-R00, dated 01 December 2011 - Instruction for Continued Airworthiness, Ref. 0496-02-A-2111-R01, dated 30 November 2011 - Aircraft Flight Manual Supplement, Ref. 0496-02-A-2301-R00, dated 05 October 2011 - Instructions for Operation Master Minimum Equipment List Supplement...</td>
</tr>
<tr>
<td>10037544</td>
<td>H4 AEROSPACE (UK) Ltd</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B767-300</td>
<td>Installation of Green Plume Seats as defined in HAAC/P016</td>
<td>JAA TCDS JAA/25/98-019 Issue 2The Certification Basis for the original product remains applicable to this certificate/app.</td>
<td>FAA STC No. ST02177LA dated 04 December 2009 Installation of Rockwell Collins Model 6605 Head-Up Display (HUD) and Enhanced Vision System (EVS) CMA-2603 Infrared Sensor System (ISS) in accordance with section 9 of the Rockwell Collins Master Drawing list 9701-1267 Revision D-1.</td>
</tr>
<tr>
<td>10037559</td>
<td>airplus engineering GmbH</td>
<td>REV. 4</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of G600 Avionics System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - The following paragraphs are applicable to this certificate/app.</td>
<td>Flight Manual Supplement PO-0972-00000, Issue 3, Description of Modification for STC Rev. 3, DE-0972-00003, Issue 1 - Engineering Work Order EW-0972-00000, Issue 5 - Maintenance Manual Supplement MM-0972-00000, Issue 3 - For later revisions of the above listed document(s) approved/accepted under this STC...</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>STC No.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>GARMIN INTERNATIONAL, Inc.</td>
<td>REV. 12</td>
<td>FAA SA02019SE-D</td>
<td>SEE FAA AML SEE ABOVE</td>
<td>Installation of Garmin GTN 6XX/7XX and GTN 7XX Series Navigation System Revision 11 adds more capabilities and new Part 23/25/26 to the AML. Revision 12 adds the EASA AML to STC 10037574 for (EASA only models).</td>
<td>The Type and OSD Certification Basis for the original product and the following: additional or alternative airworthiness requirements: See EASA Approved Annex to STC Approval dated 17/01/2012. The requirements for environmental protection and the associated certified noise and/or emissions lev...</td>
<td>- For Aircraft in FAA AML to FAA STC SA02019SE-D: Garmin Master Drawing List (MDL) 005D-533-D. Revision 1, approved 12/03/2011. The MDL identifies applicable FAA approved Airplane Flight Manual Supplements and System Maintenance Manual (including Instructions for Continued Airworthiness) The MDL, as F...</td>
</tr>
<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10037583</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>2. PA-31T, PA-31T1, PA-31T2</td>
<td>Installation of Garmin G600 System with Synthetic Vision (SVT)</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Flight Manual Supplement G600: ASR-2011-033-FMS-05-1-00 Rev. 00 Flight Manual Supplement EPUBAS: ASR-2011-033-FMS-05-2-00 Rev. 00</td>
<td>None</td>
</tr>
<tr>
<td>10037593</td>
<td>SABENA TECHNICS BOD</td>
<td>A340-542</td>
<td>STC is applicable to MSN 902 only. Note: MMEL Supplements associated with this STC are not covered by this STC approval</td>
<td>This STC is applicable to MSN 902 only. Note: MMEL Supplements associated with this STC are not covered by this STC approval</td>
<td>Modification Approval Sheet, Ref: MAS S-00-2156 Revision 6 dated 15 Sep. 2013</td>
<td>None</td>
</tr>
<tr>
<td>10037612</td>
<td>AIRBUS S.A.S</td>
<td>A320-214</td>
<td>Installation of Complete Custom Aircraft Interior - ASN 9423.</td>
<td>SAS.A.1008 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Certification Programme EASA STC No. SA04-113 Issue 1 dated 28 October 2004.</td>
<td>Restricted to ASN 9423 and 1. Refer to the Limitations sections of the required AFM Supplement and the referenced FCOM/2. WLAN System is compatible only to IEEE 802.11(b).</td>
</tr>
<tr>
<td>10037640</td>
<td>BOMBARDIER INC.</td>
<td>BD-700-1A10</td>
<td>Installation of Complete Custom Aircraft Interior - ASN 9423.</td>
<td>SAS.A.1008 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Certification Programme (including Master Data List in chapter S9UN284-001 ADC12 (1) Issue 3 dated 1st of November 2011) and later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
</tr>
<tr>
<td>10037643</td>
<td>AEROSPACE DESIGN AND CERTIFICATION LTD / AEROEROSPACE</td>
<td>A330-243</td>
<td>Installation of new Business Class Avio EOX2 Seats and IMS RAVE IFE System. The FWG cabin is currently configured with B Canal EV seats which are to be replaced with new Avio BC seats. The MAS 200E IFE system will be removed from the FWG Cabin and replaced with the IMS RAVE IFE System. There are...</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Certification Programme (including Master Data List in chapter S9UN284-001 ADC12 (1) Issue 3 dated 1st of November 2011) and later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA</td>
<td>CH#</td>
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<td>10037644</td>
<td>BOMBARDIER INC.</td>
<td>-</td>
<td>TCCA STC: SA04-113</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9429.</td>
<td>EASA.M.A.009/0009Flight System is compatible only to IEEE 802.11(b).</td>
</tr>
<tr>
<td>10037675</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC</td>
<td>No. SA05-60 ISSUE 1</td>
<td>BD-700-1A11</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9226</td>
<td>EASA.M.A.009/0009Flight System is compatible only to IEEE 802.11(b).</td>
</tr>
<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
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<td>10037677</td>
<td>AVIATION DEFENSE SERVICE</td>
<td>REV. 1</td>
<td>FAN JET FALCON BASIC</td>
<td>Carriage of ARPEGE pylons-only (Expanded Flight envelope)The purpose of this approval is to revise EASA STC 10037677 to reflectcarriage of the ARPEGE pylons only configuration on 737-200, 737-300, 737-400 or 737-500 in flightref. MDL-1259 Issue 7, MMO 0.78, FLS70, n = +2g / -1g</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Change Approval Sheet doc. No. C4-AVD-MDA20-95-POGUI Rev. 02 Issue dated 29.10.2013 Master Document List doc. No. MDA-AVD-10101 Rev.1 dated 30.11.2013 Aircraft Flight Manual Supplement doc. No. SAFM-FALC00 Rev. 101 after revisions of the above listed documents approved by EASA</td>
</tr>
<tr>
<td>10037701</td>
<td>GAMA AVIATION (ENGINEERING) LIMITED</td>
<td>REV. 3</td>
<td>CAA UK RESPONSIBILITY MODEL LIST</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>GNS-430W/530W navigation system SBAS upgrade and certification of LPVcapability.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
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<tr>
<td>10037702</td>
<td>AVMOD, LLC</td>
<td>REV. 3</td>
<td>737-400, 737-500</td>
<td>Modification of Wing Inboard and Outboard All Trailing EdgeFlaps Revision 2 of the STC with reference to EASA.IM.A.S.0116 has been issued as a result of its transfer from Aero Tech Service Inc (ATS) to AZRA LLC/ABA AeroTech Services Revision 3 of the STC has been issued as a result of its transfer from...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- AeroTech Services Inc Master Drawing List No 52-0001 Revision Cdated 20 May 2003 (for Model 737-200,-200C and -300Series Airplane)- AeroTech Services, Inc Master Drawing List No 52-0001, Revision E dated 08 January 2005 (for Model 737-400 and -500 Series airplane)- Airplane Flight Manual...</td>
</tr>
<tr>
<td>10037703</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td>A319-115</td>
<td>Airbus Corporate Jet Centre Modification CJ0800 Issue 4 - Project &quot;2A&quot;Revision due to AFM-S and MPO-S</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements as listed in the General CRI A-4305-005 &quot;AC/IC Project Certification Basis.&quot;The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>- Modification Approval Sheet ref. CJ0040 Issue 4, dated 24th October 2013- Modification Approval Sheet &quot;Conversion Private/Public and ViceVersa&quot; ref. CJ0675; - Airplane Flight Manual Supplement, ref. F&amp;MA MOD CJ0000, Revision02, dated 1st October 2013- Maintenance Planning Document - MPO - Sup...</td>
</tr>
<tr>
<td>10037704</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 1</td>
<td>A319-132</td>
<td>Introduction of Panasonic dMPES in Flight Entertainment - IFE SystemRef 1 Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval - CRI H-01 Enhanced Airworthiness Programme for Aerospace Systems - ICAA E&amp;W &quot;The requirements for environmental protection and the associated...</td>
<td>- Modification Summary, ref. doc. AES-319-1549, Issue 2, dated 14thDecember 2011- Master Data List, ref. doc. AES-319-1549ML, Issue 2, dated 14thDecember 2011 or later revisions of the above listed documents approved by EASA</td>
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<td>EASA STC No.</td>
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<td>STC Type</td>
<td>STC No.</td>
<td>Rev. Level</td>
<td>Description</td>
<td>Certification Basis</td>
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<td>10037705</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA STC NO. SA11-89 ISSUE 1</td>
<td>CL-600-2B16</td>
<td>Modification to Potable Water Tank Air Supply</td>
<td>EASA A.023 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. SA11-89 Issue 1 dated 07 November 2011. ACS-NAI Ltd. Modification Data Summary Number H311000 Revision NC or later approved revision. Instructions for Continued Airworthiness.</td>
</tr>
<tr>
<td>10037710</td>
<td>ACS-NAI Ltd.</td>
<td>TCCA STC C-LSA10-169/D</td>
<td>CL-600-2B19</td>
<td>Installation, Complete Custom Aircraft Interior.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. C-LSA10-169/D Issue 1 dated 03 December 2010. ACS-NAI Ltd. Modification Data Summary G773000 Revision NC or later approved revision.</td>
</tr>
<tr>
<td>10037711</td>
<td>ACS-NAI Ltd.</td>
<td>TCCA STC C-LSA11-165/D</td>
<td>CL-600-2B19</td>
<td>Installation, Complete Custom Aircraft Interior.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCCA STC No. C-LSA11-165/D Issue 1 dated 11 October 2011. ACS-NAI Ltd. Modification Data Summary H130000 Revision NC or later approved revision.</td>
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<td>10037721</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A330-341 / -342 / -343</td>
<td>A350-321 / -322 / -323</td>
<td>Thai Airways A330-300 - Seat exchange and electrical adaption. This STC installs 263 Y class seats, with associated electrical adaptations.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>TCD5 EASA A.025 - CS-25 Appendix H25.5 (amdt 8) - CRI F-3302-002 : IFE and IPSS. The requirements for environmental protection and the...</td>
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<tr>
<td>10037753</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>A300F4-608ST</td>
<td>A300F4-608ST</td>
<td>Special Transports / Transports Speciaux Issue 1 of ATI STC ref. ST25MD2020 (Applicant reference)</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>EASA TCD5 A.014 - CRI F-606 Issue 2 : Transport of Instrumental Packages powered by an aircraft electrical supply. The requirements for...</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td>1900, 1900C, 1900D, 2000C</td>
<td>Installation of Garmin 500 Series system with coupling to auto/flight-guided Collins FCS 80, according to SEAM modification file no. 3541/11</td>
<td>Flight Manual Supplement no. SEAM-3541-AMF, revision 4 dated 09/12/2011 STC Data List no. SEAM-3541-SDL, revision 6 dated 26/12/2011</td>
<td>Installation of this design change must be performed by SEAM technical personnel if the GNI-500 Series system is already installed on the airplane, a third independent radiocommunicati and/or navigation system, not similar to GNS Systems, must also be in...</td>
<td>22/12/2011</td>
<td>Active</td>
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The Certification Basis for the original product remains applicable for the installation of Brite Saver Usage Monitoring System Rev. 1. Removal of the limitation “only for EC135/E635 variants with Cockpit” Rev. 2. Change STC certificate holder name Rev. 3. Extension of the applicability to the EC135 P3 EC635 P3 and EC135 T3 models Rev. 4. Software evolution in...
<table>
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<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA Code</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tbody>
<tr>
<td>10038022</td>
<td>ACS-NAI Ltd.</td>
<td>TC A 3</td>
<td>NO. C- LSA1-0081D</td>
<td>CL-600-2B19</td>
<td>Installation, Complete Custom Aircraft Interior.</td>
<td>EASA MA-A.0237 Certification Basis for the original product and applicable techis certificate approval.</td>
<td>TCCA STC No. C- LSA1-0091 Issue 2 dated 22 July 2011 ACS-NAI Ltd. Modification Data Summary</td>
<td>Restricted to ARN 81051-The Wireless Local Area Network (WLAN) system installed as part of an aircraft completion is intended to provide cached internet content and email services to Portable Electronic Devices. Any other use of this equipment will require a re-examination of the certification basis.</td>
<td>12/01/2012</td>
<td>Active</td>
<td>10038022</td>
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<tr>
<td>10038053</td>
<td>FOKKER SERVICES B.V.</td>
<td>F28 MARK 0070</td>
<td></td>
<td>HEARJET 45 46</td>
<td>Rev. 1: Installation of Air Ambulance Equipment Reason for Rev. 1: added configuration for Adiposity patients.</td>
<td>The Certification Basis for the original product and the following:</td>
<td>-CRF-21 Wireless Local Area Network(WLAN)- Certification Plan CP-0119 Issue 2-APM change AEROMET2011-007 Issue 10/2011</td>
<td>-Installation of EASA STC 10027637 is a prerequisite for this STC-Limitation of Wireless Local Area Network as documented in APM changeAEROMET2011-007</td>
<td>16/01/2012</td>
<td>Active</td>
<td>10038053</td>
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<tr>
<td>10038064</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>LEARJET 45</td>
<td>46:LEARJET 45</td>
<td>REV. 1</td>
<td>Installation of Air Ambulance Equipment Reason for Rev. 1: added configuration for Adiposity patients.</td>
<td>The Type and AvS Certification Basis for the original product and any amended by the following:</td>
<td>-AEROLITE LJ45-19010-050-LP, AEROLITE LJ45-25-20-45-000-505-2, 507-2 Rev. 8Upd. Nov 30, 2023 - Aeroplane Flight Manual Supplement AAM-AAT-59-LH45-EMS Rev. 2</td>
<td>Per Aeroplane Flight Manual Supplement AAM-AAT-59-LH45-EMS No medical equipment is part of this approval. - Minor Change LR46-1010-050-LP, Installation of EMS power source module installed prior to this modification: AEROLITE EMS Kit, Installation of EMS power source EASA STC 10046038, must be installed.</td>
<td>05/12/2023</td>
<td>Active</td>
<td>10038064</td>
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<tr>
<td>10038070</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>A330-2431-341-343</td>
<td></td>
<td>REV. 1</td>
<td>This modification intends to:- Installing the Ball-Mount support on the cockpit sliding windows: Install the two Pilot Terminal Units (PTU) in the cockpit on the Ball-Mount support: Install OFWCU thin mounting box: Install AMBDP 3MCU fixed mounting box in the E/E bay: Install AMDPB and QWCUW equip...</td>
<td>The Certification Basis for the original product and the following:</td>
<td>Certification Programme (including Master Data List in chapter 9) AEI15811PCD Instructions for Continued Amortiveness AEI15811AFCFlight Manual Supplement AEI15811MFr/W</td>
<td>None</td>
<td>05/06/2012</td>
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<td>10038071</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>767-31K ER</td>
<td>Installation of Geven 'Novel &amp; Puma' Series Passenger Seats 326 configuration.</td>
<td>EASA CS25 Change 4 The Certification Basis for the original product and the following:</td>
<td>Modification Summary Sheet and Declaration of Compliance HH41389 Issue 1 dated 04/01/12 or later revisions of the above listed documents approved by EASA.</td>
<td>Limited to installation on B767-31K Serial Numbers 27205, 27206 and 28865.</td>
<td>16/01/2012</td>
<td>Active</td>
<td>10038071</td>
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<td>10038077</td>
<td>TELAIR INTERNATIONAL, LLC</td>
<td>FAA STC 7003856A</td>
<td>747-8</td>
<td>Installation of a Lower Deck Cargo Handling System</td>
<td>The Certification Basis for the original product and the following:</td>
<td>FAA STC Number S7003856A dated 23.12.2011 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>17/01/2012</td>
<td>Active</td>
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<td>10038094</td>
<td>DUNLOP AIRCRAFT TYRES LIMITED</td>
<td>REV. 1</td>
<td>ICAO</td>
<td>CL-600-2B19 (RU SERIES 100)</td>
<td>Installation, Dunlop Aircraft Tyres Ltd P/N DR279217T Main and DR151127Nose Gear Tyres. This STC is issued in order to reflect the transfer of ownership from KCS-NAI Ltd to Dunlop Aircraft Tyres Ltd. Rev. 1. Update of technical documentation. This STC has previously been validated in the following European...</td>
<td>- Aeronautical Services, Modification Data Summary C005506s rev.000 DATL PR86 MDL Rev. R - Dunlop Aircraft Tyres Ltd. Document Number DM1172 General Servicing Instructions - Instructions for Continued Airworthiness KT5690060 Rev IR issued as DATL PR-06-ICA Rev Bor later revisions of the above.</td>
<td>Different nose landing gear part number lists may not be intermixed on the same axle. Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/roughness effect...</td>
<td>12/04/2016</td>
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<td>10038094</td>
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<tr>
<td>10038098</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>REV. 1</td>
<td>TCCA</td>
<td>DHC-8-202</td>
<td>TCDA STC D-LSA06-139/D</td>
<td>FAR 25 as per EASA M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation must be in accordance with TCCA approved FieldAviationEast Ltd. Master Drawing and Data List DDL-1158, Rev. C dated 15 August 2011. Maintenance must be in accordance with TCCA accepted Maintenance Manual Supplement 20169, Rev. A dated 08 July 2011 or later revisions of the above listed.</td>
<td>The modification is limited to Serial numbers 335, 337, 445, 446.</td>
<td>19/01/2012</td>
<td>Active</td>
<td>10038098</td>
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<tr>
<td>10038103</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>REV. 2</td>
<td>FAA STC</td>
<td>EC135 T2</td>
<td>FAA STC SR01042SE</td>
<td>Fabrication of Dart D135-692-011 Bearpaws in accordance with Dart Aerospace Drawing List No. MDL-135-692 and installation of this system.</td>
<td>- 1. Flight Manual Supplement FMS-D135-692-EASA Revision A or later revisions of the above listed documents approved by EASA. 2. Dart Aerospace USA, Inc. Master Document List No. MDL-D135-692, Revision H dated 06 June 2017 or later FAA approved revision. 3. Installation Instructions Document No. (IE)135-6...</td>
<td>1. Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/roughness effect upon the aircraft's thrust and/or emissions level of the original product. 2. Approval of this change in type design approval...</td>
<td>21/08/2017</td>
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<tr>
<td>10038104</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>TCCA</td>
<td>DHC-8-202</td>
<td>TCCA STC D-LSA06-111/D</td>
<td>FAR 25 as per EASA M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation must be in accordance with TCCA approved FieldAviationEast Ltd. Master Drawing and Data List DDL-1158, Rev. C dated 22 July 2011. Maintenance must be in accordance with TCCA accepted Maintenance Manual Supplement 20169, Rev. B dated 22 July 2011 or later revisions of the above listed...</td>
<td>The modification is limited to Serial numbers 445, 446.</td>
<td>19/01/2012</td>
<td>Active</td>
<td>10038104</td>
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<td>10038105</td>
<td>STANDARD AERO</td>
<td>FAA STC</td>
<td>ST02912CH-D</td>
<td>MF50</td>
<td>Installation of an interior refurbishment and a pocket door/Validation of FAA STC N° ST02912CH-D</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Standard Aero Master Data List doc. N° 1016814 Revision B dated January 08, 2011. Instructions for Continued Airworthiness doc. N° 1016827 Revision B dated January 09, 2011. FAA STC N° ST02912CH-D issued dated January 09, 2011 or later revisions of the above listed.</td>
<td>None</td>
<td>19/01/2012</td>
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<td><strong>10038108</strong></td>
<td>AIRCRAFT TECHNICAL DEVELOPMENT Inc.d/b/a AEROCON ENGINEERING COMPANY</td>
<td>Cabin Interior Noise Reduction Systems installation kit in Bombardier CL-600-2C10 (Regional Jet Series 700, 701 &amp; 702) Series Aircraft</td>
<td>EASA M.A.0237 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificates of noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. -FAA STC No. ST02828-A dated 16 November 2011 -Aerocron Engineering Company Master Drawing List No. A225- NRC/CL600-1000F (Revision B dated 29 October 2011 or later approved revision) -Aerocron Engineering Company Maintenance Manual Supplement Document No. MM5-CL600-168B Revision A dated 28 October 2011 or later.</td>
<td>None</td>
<td>19/01/2012</td>
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<td><strong>10038123</strong></td>
<td>AEROCONTROLE X GROUP, INC.</td>
<td>Installation of an AeroControlex Potable Water Level Sensor/Assembly P/N 8430-00000</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval - CRH-1/1CA on EWS 8430-00000/DL Rev. Original, no-date 842/15 Installation Instructions Rev. C. dtd. 27 Jan 2012-ER-10-08 Instructions for Continued Airworthiness Rev. 2. dtd. 18 Jun 2012-ER-10-09 Instructions for Continued Airworthiness Rev. 2. dtd. 18 Jun 2012-ER-10-10 Instructions for Continued Airworthiness Rev. 2. dtd. 18 Jun 2012</td>
<td>None</td>
<td>20/01/2012</td>
<td>Active</td>
<td>10038123</td>
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<td><strong>10038130</strong></td>
<td>HAEKO CABIN SOLUTIONS, LLC</td>
<td>Installation of LCD Monitor Upgrade</td>
<td>TIMCO MDL 10/74/01, Revision F, 24/05/2011 Instructions for Continued Airworthiness 107024R009, Revision C, 16/05/2011107024R009, Revision IR, 26/09/2011 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/034/CF or subsequent revisions of this document.</td>
<td>None</td>
<td>21/06/2012</td>
<td>Active</td>
<td>10038130</td>
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<td><strong>10038134</strong></td>
<td>BELL TEXTRON, Inc.</td>
<td>Replacement Windows</td>
<td>AA05303 revision M dated 20 September 2010 - Drawing List and Installation Instructions.</td>
<td>None</td>
<td>26/01/2012</td>
<td>Active</td>
<td>10038134</td>
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<td><strong>10038135</strong></td>
<td>HAWKER BEECHCRAFT Ltd.</td>
<td>To introduce Dual Li3 Electronic Flight Bags (EFB) units in the cockpit/flight power supplies.</td>
<td>- Installed to Major Change SCO362A at issue 4 or later-Hawkex Beechcraft Limited Aircraft Flight Manual Supplement 3 to AFM140-58033-0005 (Hawker 750), or Supplement 3 to AFM 140-58033-005(Hawker 801XP), or Supplement 2 to AFM 140-58053S-0005 (Hawker 850XP) or Supplement 2 to AFM 140-58083S-0005 (Hawker 850XP) or Supplement 3 to AFM 140-58033T-005 (Hawkex). All EFB functionality and software approval will be the responsibility of the aircraft operator. Unless approved otherwise by all Required Manuals (Hard Copy) must still be carried in the aircraft.</td>
<td>All EFB functionality and software approval will be the responsibility of the aircraft operator. Unless approved otherwise by all Required Manuals (Hard Copy) must still be carried in the aircraft.</td>
<td>23/01/2012</td>
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<td>STC Holder</td>
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<td>ACS-NAI Ltd.</td>
<td>Modifications to Existing Interior Completions -ASN 9393.</td>
<td>TCCA STC No. LS011-179D</td>
<td>Restricted to ASN 9393 and The Installation of EASA STC No. 10036060 is a prerequisite for the approval of this STC.</td>
<td>24/01/2012</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>Atlitalia Cabin Reconfiguration</td>
<td>TCCA STC No. LS301-1050</td>
<td>Change approval sheet CAS-B777-11-003 Rev 0 dated 19 Jan 12or later revisions of the above listed documents approved by EASA</td>
<td>24/01/2012</td>
<td>Active</td>
<td>10038157</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>Installation of 5-blades MTV-27-1-E-C-F-R (F)YT5210-58d propellers</td>
<td>EASA STC No. SH5006SW</td>
<td>None</td>
<td>09/01/2013</td>
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<td>10038174</td>
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<td>TECH-TOOL PLASTICS, Inc.</td>
<td>Installation of Tech-Tool performed acrylic plastic windshield and nose cone with Tech-Tool Drawing List (Doc AS351-2023) No 1, Rev A, dated 08/28/09 (FAA STC SH5022SW)</td>
<td>FAA STC SH5022SW</td>
<td>None</td>
<td>26/01/2012</td>
<td>Active</td>
<td>10038180</td>
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<td>TECH-TOOL PLASTICS, Inc.</td>
<td>Installation of Tech-Tool performed acrylic plastic windshield and nose cone with Tech-Tool Drawing List (Doc AS353-2023) No 1, Rev A, dated 08/28/09 (FAA STC SH5022SW)</td>
<td>FAA STC SH5022SW</td>
<td>None</td>
<td>26/01/2012</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>Installation of wedge Windows with Sliders into LH and RH Passenger Doors in accordance with Aeronautical Accessories Inc. Installation Instructions Report Number AA-0004, Revision A, 09. (FAA STC SR02808NY)</td>
<td>FAA STC SR02808NY</td>
<td>None</td>
<td>26/01/2012</td>
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<td>10038194</td>
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Note: The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate/approval or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.
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<th>STC Holder/Description</th>
<th>FAA STC/STC ref.</th>
<th>STC Models</th>
<th>Description</th>
<th>Certification basis</th>
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<tr>
<td>10038222</td>
<td>FAA SA01696LA</td>
<td>EMB-145ER/EU/EP/MP/REV. 1</td>
<td>Installation of Hawker Part Number 9750W0370 or 9750W0370-85Series/4Volt 43 Ah Sealed Lead Acid Battery.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged. It must be determined the interaction between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td></td>
<td>Prior to installation of this modification it must be determined the interaction between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/11/2014</td>
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<td>10038222</td>
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<tr>
<td>10038231</td>
<td>FAA STC SR00707DE</td>
<td>429</td>
<td>Automatic Door Openers for Crew and Passenger Doors and Pilot and/or Pilot Floor Protectors</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged. It must be determined the interaction between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>FAA-09025 revision B dated 6 November 2009 - Drawing List for Door Openers, Floor Protectors and Paint Protectors; AA-09046 revision B dated 7 July 2010 - Installation Instructions, Crew Floor Protector K2-2A-09070</td>
<td>None</td>
<td>30/01/2012</td>
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<td>10038231</td>
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<td>10038232</td>
<td>FAA STC SR00692DE</td>
<td>429</td>
<td>Installation of Bleed Air Heater</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged. It must be determined the interaction between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>429H-1 Issue dated 8 July 2011 - Flight Manual</td>
<td>None</td>
<td>30/01/2012</td>
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<td>10038234</td>
<td>FAA STC SR01696LA</td>
<td>EMB-145ER/EU/EP/MP/REV. 1</td>
<td>EASA Validation of ANAC STC 2005/010-18. Electronic Flight Bag (EFB) Installation Description as per ESCMC002G-Rev A, dated 17th January 2012.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged. It must be determined the interaction between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>ESMC002G-Rev A, dated 17th January 2012.</td>
<td>None</td>
<td>01/02/2012</td>
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<td>10038238</td>
<td>FAA STC SA02387AK</td>
<td>2- F33A, F33C</td>
<td>Fabrication and Installation of Exterior LED Lighting Suite in accordance with FAA STC SA02387AK.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged. It must be determined the interaction between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>2- F33A, F33C</td>
<td>Prior to installation of this design change it must be determined the interaction between this modification and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/01/2013</td>
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<tr>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>TCGA- S6942-9</td>
<td>REV. 1</td>
<td>AS 355 N, AS 355 NP F1, AS 355 F2, AS 355 E, AS 355 F, AS 355 B, AS 355 B1, AS 350 BA, AS 350 D, AS 362 B2, AS 360 B3</td>
<td>Leit-Man Litter Kit installed as per Master Drawing List No. MLD-99-011 Revision 1. Changed STCch reference and updated documentation referencing modified compatibility table the MLD.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged; and remain applicable to this certificate/approval without any impact.</td>
<td>Master Drawing List No. MLD-99-011 issued 2022.</td>
<td>Prior to installation of this change/repair it must be determined that the airworthiness of the product.</td>
<td>09/02/2020</td>
<td>Active</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>F80 STC ST02890NY-D</td>
<td>REV. 3</td>
<td>FAA STC ST02890NY-D</td>
<td>A321-212, -213, -232 A3201-111, -112, -131, -211 A320-211/212/213 A320-211/212-214 A319-151/152-153 A319-151-152-153-114-115</td>
<td>Installation of Panasonic Global Communications Suite (GCS) antenna. This modification installs a large antenna on top of the fuselage, isolating the Panasonic GCS system from passengers and providing clear line of sight for aircraft operations (without coupling of GPS). This installation is not similar to the GNS 400 radionavigation equipment with at least one person.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC ST02890NY-D amended December 21, 2013 Master Data List Document, ref: 0164-07139-2268, Revision 0, dated August 20, 2012.</td>
<td>Prior to installation of this design change must be performed by authorized SEAM technical personnel. The aircraft must be equipped with at least one radionavigation transponder with Elementary Surveillance Mode S (without coupling of GNSS).</td>
<td>02/09/2015</td>
</tr>
<tr>
<td>HAWKER BEECHCRAFT LTD (EASA.A.S.03569)</td>
<td>FAA STC ST02890NY-D</td>
<td>REV. 1</td>
<td>FAA STC ST02890NY-D</td>
<td>A3201-212, -213, -232 A320-211/212/213 A319-151/152-153 A319-151-152-153-114-115</td>
<td>Installation of Panasonic Global Communications Suite (GCS) antenna. This modification installs a large antenna on top of the fuselage, isolating the Panasonic GCS system from passengers and providing clear line of sight for aircraft operations (without coupling of GPS). This installation is not similar to the GNS 400 radionavigation equipment with at least one person.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC ST02890NY-D amended December 21, 2013 Master Data List Document, ref: 0164-07139-2268, Revision 0, dated August 20, 2012.</td>
<td>Prior to installation of this design change must be performed by authorized SEAM technical personnel. The aircraft must be equipped with at least one radionavigation transponder with Elementary Surveillance Mode S (without coupling of GNSS).</td>
<td>01/02/2012</td>
</tr>
<tr>
<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINES A.R.L</td>
<td>F177RG</td>
<td>REV. 1</td>
<td>F177RG</td>
<td>Installation of a Garmin GNS 400 series system, a Garmin GNS 400S antenna, a Mid-Continent MD41 annunciator and a Garmin GTX 330 transponder.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>SEAM Flight Manual Supplement No. 3623/11, revision 0 dated 2012/01/01, SEAM Modification File No. 3623/11, revision 2 dated 2012/01/01, or later revisions of the above listed documents approved by EASA.</td>
<td>Installation of this design change must be performed by authorized SEAM technical personnel. The aircraft must be equipped with at least one radionavigation transponder with Elementary Surveillance Mode S (without coupling of GNSS).</td>
<td>01/02/2012</td>
<td>Active</td>
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<tr>
<td>AVNCS AVIATION ITALIA S.P.A</td>
<td>CL-215-6B11 (CL-415 VARIANT)</td>
<td>REV. 3</td>
<td>CL-215-6B11 (CL-415 VARIANT)</td>
<td>Installation of a Garmin GNS 400 series system, a Garmin GNS 400S antenna, a Mid-Continent MD41 annunciator, a Garmin GTX 330 transponder, and a Mid-Continent MD41 annunciator.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List Doc. No. MLD-34-59-0912-A Rev. 0 dated 19 January 2012.- Aircraft Flight Manual Supplement Doc. No. FMS-34-59-0912-A Rev. 0 dated 09 January 2012.- Minimum equipment list MEL-34-59-0912-A Rev. 0 dated 09 January 2012.-</td>
<td>Prior to installation of this change/repair it must be determined that the airworthiness of the product.</td>
<td>15/06/2023</td>
<td>Active</td>
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<td>STC Holder</td>
<td>FAA STC No.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>THALES AVIONICS Inc.</td>
<td>ST0411LA</td>
<td>In-Flight Entertainment (IFE) System Installation</td>
<td>FAA STC No. ST0411LA, dated 20 January 2012, Thales Avionics Index List, Drawing No. 254709, Revision &quot;A&quot;, dated December 16, 2011, or later approved revision. The instructions for Continued Airworthiness are called out in the associated Master Drawing Lists (MDL) defined in the above referenced Index L..</td>
<td>Concurrent installation of the following STCs are required for the installation of this STC: 1) Flight Structures, Inc. EASA STC No. 10038267, Business Class Seat Installation 2) Northwest Aerospace Technologies, Inc. EASA STC No. 10038268 (EASA Project No. 0010011523), Aircraft Interior Reconfiguration...</td>
<td>03/02/2012</td>
<td>Active</td>
<td>10038267</td>
<td></td>
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<tr>
<td>NORTWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>ST0412LA</td>
<td>First and Economy Class Cabin Interior - BA 777-200 Aircraft</td>
<td>FAA STC No. ST0412LA, dated 18 January 2012, Northwest Aerospace Technologies, Inc. (NAT) Master Drawing List No. 836-1210; Rev &quot;C&quot;, dated December 13, 2011 or later approved revisions. Instructions for Continued Airworthiness Document No. M365-270; Rev &quot;C&quot;, dated September 29, 2011 or later approved.</td>
<td>Concurrent installation of the following STCs are required for the installation of this STC: 1) Flight Structures, Inc. EASA STC 10015720, Business Class Seat Installation 2) Thales Avionics, Inc. EASA STC No. 10038267 (EASA Project No. 0010011945), Passenger Cabin IFE System Installation, and 3) Northwe...</td>
<td>02/02/2012</td>
<td>Active</td>
<td>10038286</td>
<td></td>
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<tr>
<td>ICELANDAIR ENGINEERING DEPARTMENT</td>
<td>B757-200</td>
<td>Installation of Satcom and Flight Tracking System.</td>
<td>The Certification Basis for the original product remains applicable to this approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this approval.</td>
<td>None.</td>
<td>02/02/2012</td>
<td>Active</td>
<td>10038271</td>
<td></td>
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<tr>
<td>RAM AIRCRAFT, LP</td>
<td>SA0469SC-D</td>
<td>Modification of Cessna 414 in acc. with FAA STC SA04699SC-D: Aircraft modified in accordance with RAM Master Drawing No. 1982, Revision K, FAA approved, September 23, 2009 and Installation of modified TCM engine model TSIO-520-NB in accordance with FAA STC SE09469SC-D or SE09104SC-D.</td>
<td>The Certification Basis for the original product remains applicable to this approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this approval.</td>
<td>This approval is limited to the installation of this STC: 1) Foreign NOTAMs 10110209, 2) PHYS-0010011945, 3) MFD Video functionality (fixed video image) has not...</td>
<td>03/02/2012</td>
<td>Active</td>
<td>10038295</td>
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<tr>
<td>LEARJET Inc.</td>
<td>ST029680</td>
<td>Installation of a Honeywell DU-875 or DU-875AF Liquid Crystal Display(LCD) based Display System including two Cursor Control Devices (CCDs)</td>
<td>The Certification Basis for the original product remains applicable to this approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this approval.</td>
<td>None.</td>
<td>16/10/2020</td>
<td>Active</td>
<td>10038289</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA</td>
<td>Doc.#</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10038288</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TC1A STC</td>
<td>SA11-88</td>
<td>CL-600-2B16</td>
<td>Electronic Flight Bag Class II (PED) (Provisions)</td>
<td>- TCCA STC No. SA11-88 Issue 1, dated 25 October 2011 - ACS NAA Ltd. Modification Data Summary Number H20003 Revision N/C dated 25 October 2011 or later approved revision - ACS NAA Ltd. Airplane Flight Manual Supplement H200090 Revision N/C or later approved revision - Bombardier Airworthiness Limited...</td>
<td>None</td>
<td>03/02/2012</td>
<td>Active</td>
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<tr>
<td>10038291</td>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY INC USA INC</td>
<td>REV.</td>
<td>FAA STC</td>
<td>ST00011B</td>
<td>EMB-135BJ</td>
<td>Installation of a cabinet and a 17” video monitor on EMB-135BJ aircraft/No/This STC is revised at Rev. 1 in order to record the transfer of the original ANAC STC No. 2011500-02 to a United States based Design Approval Holder.*</td>
<td>-Master Document List N° ESSC0015 Rev. 0, dated 4 December 2018 - Airplane Flight Manual Supplement doc. N° H. 02-3679-U/ANAC/2011 Rev. B dated 20 December 2018-7 for Continued Airworthiness (ICA), document numberESCMTD009 Rev. B, dated 4 December 2018- FAA STC No. ST00011B, released on 4 No...</td>
<td>None</td>
<td>05/01/2021</td>
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<tr>
<td>10038292</td>
<td>SOFINARE SOCIEDADE DIFINACIAMENTO DE REPRESENTACAO ES SA</td>
<td></td>
<td>TR182</td>
<td>R182</td>
<td>Installation of Integrated Flight System Garmin G-500 and DoubleNav/Com/GPS systems GNS-430W</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval without any im...</td>
<td>None</td>
<td>03/02/2012</td>
<td>Active</td>
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<tr>
<td>10038293</td>
<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td></td>
<td>BN-2</td>
<td></td>
<td>BEZU installation on a BN-2 rev.2 du 27/01/2012.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The certificate/ not applicable to the original product and remain applicable to this certificate/ approval.</td>
<td>None</td>
<td>03/02/2012</td>
<td>Active</td>
</tr>
<tr>
<td>10038294</td>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV.</td>
<td>A310-300</td>
<td></td>
<td>Amori Flight A310-300 Cabin Refurbishment upgrade the existing cabin by... Removal of the existing seats and installation of new seats in the, Business and Economy class. Replace the existing IFE system- Installation of a new VCC class dividers, dog house and decor panels. Installation of...</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>- Master Document List, Doc. No. S.OSM.00003-MDL, Rev. 0-AFMS, Doc. No. S.OSM.00003-AFMS, Rev. 0-ICA, Doc. No. S.OSM.00003-ICA, Rev. 00 later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>02/04/2021</td>
</tr>
<tr>
<td>10038296</td>
<td>AVIONES, PIEZAS Y ACCESORIOS, S.L.</td>
<td>REV.</td>
<td>DQAC-E 138</td>
<td>B. 206B, 206L A. 8206 B</td>
<td>Transfer EASA STC 10038206 - Transfer from INARER MAINTENANCE S.A.U. to INARER, PIEZAS Y ACCESORIOS Power Lines cleaning system with high pressure water with form &quot;DGAC-ECertificateData Sheet No. 138&quot; is composed by a reservoir tank, a high pressure power system, automatic lance to eject thes...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement (Bell 206B series &amp; Agustawestland AB206- Bizweseries) INARER.1200409-25-FMS-1 R3 Flight Manual Supplement (Bell 206L-3 &amp; L-4) INARER.1200409-25- FMS-2 R2 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>06/05/2013</td>
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<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>BELL TEXTRON, Inc.</td>
<td>FAA STC:SR09534RC-D</td>
<td>BELL 429</td>
<td>Installation of Cargo Mirror</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>-STC-AT-030 remaining Original dated 8 April 2010 - Rotorcraft Flight Manual Supplement later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions, if applicable) STC-AT-030</td>
<td>None</td>
<td>06/02/2012</td>
<td>Active</td>
<td>10038305</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>P.68TC &quot;OBSERVER&quot;</td>
<td>P.68 &quot;OBSERVER 2&quot;</td>
<td>Electric Camera Lift installation/ Installation of an electric camera lift, an operator console and acamera lift controller</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>- Change Dossier &quot;Electric Camera Lift&quot; Doc. No. STC-AT-038, Rev. 1 - Installation instructions Doc. No. 038- II-STC-AT, Rev. 0 - Flight Manual Supplement AFM-STC-AT-038, Rev. 1 - Maintenance Manual Supplement AMM-STC-AT-038, Rev. 1 - Installation instructions Doc. No. 038-II-STC-AT, Rev. 0 - Flight Manual Supplement AFM-STC-AT-038, Rev. 1 - Maintenance Manual Supplement AMM-STC-AT-038, Rev. 1 -</td>
<td>None</td>
<td>06/02/2012</td>
<td>Active</td>
<td>10038317</td>
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<tr>
<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td>500, 550, 560, 560XL, S550</td>
<td>Installation of Garmin Mode A/C/S Transponder GTX 330D with an additional diversity antenna</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any improvements.</td>
<td>- Master Document List: MD - 139 - 00000 rev 4 - Flight Manual Supplement; PG - 139 - 00000 rev 3 - Maintenance Manual Supplement; MM - 139 - 00000 rev 3 - Weight &amp; Balance CA - 139 - 00002 rev 1 - Aircraft Model List; ML - 139 - 00000 rev 3 - Maintenance Manual Supplement AMM-STC-AT-038, Rev. 1 - Installation instructions Doc. No. 038-II-STC-AT</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect on the airworthiness of the product.</td>
<td>10/02/2021</td>
<td>Active</td>
<td>10038318</td>
</tr>
<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>P2068T</td>
<td></td>
<td>Hatch Cover.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installed in accordance with Supplemental Type Certificate STC-AT-037, Rev. 2 Maintained in accordance with AMM Supplement AMM-STC-AT-037, Rev. 2 -</td>
<td>STC-AT-003 must be installed prior to this modification.</td>
<td>07/02/2012</td>
<td>Active</td>
<td>10038324</td>
</tr>
<tr>
<td>S-TEC CORPORATION</td>
<td>FAA STC:SA8392SW-D</td>
<td>PA-32R-301, PA-32R-301T, PA-32-301, PA-32-301T</td>
<td>Installation of S-TEC system 55/55 Two Axis Automatic Flight Guidance System, Model ST-634 (14V - System)</td>
<td>The Certification Basis of the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA approved S-TEC Bulletin No. 634, Revision 4, dated 16-Mar-2011, original FAA approved revision FAA approved S-TEC Master Drawing List No. 92683, Revision 2, dated 16-Mar-2011, original FAA approved revision FAA / DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N B...</td>
<td>Installer must check availability of EASA-approved FAA or one or more STCs when the airplane is modified per that/ those STC(s)</td>
<td>07/02/2012</td>
<td>Active</td>
<td>10038329</td>
</tr>
<tr>
<td>AVIONICS GROUP INC.</td>
<td>FAA STC:ST01595AT</td>
<td>767-200, -300, -300F, -400ER</td>
<td>Installation of Honeywell (TRA-67A) or Collins (TPR901) Mode S Enhanced Surveillance System</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Avionics Group MDL, Ao-34-1223-MDL, Revision B. 12/03/2000/AM14, Revision IR, 28/03/2000/ir later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions, if applicable) STC-AT-030</td>
<td>None</td>
<td>08/02/2012</td>
<td>Active</td>
<td>10038333</td>
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<tr>
<td>STC Holder</td>
<td>FAA STC Ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis for the original product remains applicable to this certificate/ approval</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AIRCRAFT CABIN MAINTENANCE GmbH</td>
<td>A318, A319, A320, A321</td>
<td>Installation of refurbished Passenger Seat Belt Pin EPAD2011-1-02-33 on 16g seats</td>
<td>CS-25 The Certification Basis for the original product remains applicable to this certificate/ approval</td>
<td>Refer to Master Document List E/O No. 62-146169 Rev. 0 (16.12.2011)or later revisions of the above listed documents approved by EASA</td>
<td>As per Service Bulletin No. 62-146169 SEB 06 00, latest revision.</td>
<td>08/02/2012</td>
<td>Active</td>
<td>10038356</td>
<td></td>
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<tr>
<td>TELEDYNE CONTROL, LLC</td>
<td>FAA STC 1013030</td>
<td>125-700A, 125-800A, 125-400A, 125-600A</td>
<td>Replacement of existing batteries with Teledyne GIL-036E(BB3837E)(Series Lead Acid Type Batteries)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification notice and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Drawing No 0105-57, and/or Drawing No 3702. Drawing Index, no revision and Installation Instructions dated November 21, !979. Drawing Control Sheet 3700-82, dated May 6, 1985. Operation in accordance with FAA Approved Airplane Flight Manual Supplement dated Sept.</td>
<td>08/02/2012</td>
<td>Active</td>
<td>10038359</td>
<td></td>
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<tr>
<td>MAX-VIZ Inc.</td>
<td>FAA STC 103834</td>
<td>C90GTI C90GT1 B300/C B300GTI B200/CT B200C/B200CT/ B200T</td>
<td>Installation of the Max-Viz® Enhanced Vision System (EVS) Model-1000 on Beechcraft Corporation B200/B300/ C90 Series aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Installation in accordance with One Sky Aviation, LLC Master Data List, Doc No 20 11 00, Rev 6, dated September 05, 2012. Operation in accordance with Aircraft Flight Manual Supplement, One Sky Aviation, LLC document number 32 11 04, Rev C, dated July 09, 2012, for Continued Airworthiness...</td>
<td>16/08/2016</td>
<td>Active</td>
<td>10038364</td>
<td></td>
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<tr>
<td>AERO TWIN, Inc.</td>
<td>FAA STC 103836</td>
<td>208, 208B</td>
<td>Fabrication and Installation of Aero Twin, Inc. Exhaust Deflector P/NED8-100 in accordance with FAA STC 1013030.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification notice and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/ approval.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce any adverse effect upon the airworthiness of the product.</td>
<td>09/02/2012</td>
<td>Active</td>
<td>10038365</td>
<td></td>
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<tr>
<td>IMAGIK INTERNATIONAL AERIAL REPAIR STATION, CORP.</td>
<td>FAA STC NO. 103838</td>
<td>BOEING 767-200-300</td>
<td>Installation of IMAGIK LCD Entertainment Monitors.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification notice and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA STC No. ST03404AT dated 25 February 2008, Airbus Support Group Master Data List Drawing 26012-MLD, Revision 0 dated 15 May 2007 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/C (or subsequent revisions off this decision)</td>
<td>09/02/2012</td>
<td>Active</td>
<td>10038366</td>
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<tr>
<td>JET AVIATION AG, BASEL</td>
<td>FAA STC NO. 103839</td>
<td>A319-133</td>
<td>VIP Cabin Interior Completion/See Master Data List BCJ44151-001</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.</td>
<td>FAA Flight Manual Supplement BCJ44151-001-01 Rev. B, Instructions for Continued Airworthiness BCJ44151-150-01 IPE/WS/Instructions for Continued Airworthiness.</td>
<td>15/02/2012</td>
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<td>STC Holders</td>
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<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 2</td>
<td>PA-46-500TP, PA-46R350T, PA-46-350P</td>
<td>Installation of HD Xeron LOG/TAXI and LED NAV Light</td>
<td>EASA STC 10039427 from NAYAK Aircraft Service GmbH based on NAYAK PA46-NY-09199-MDL-to be operated in accordance with NAYAK AFMS PA46-NY-09199-08-to be maintained in accordance with NAYAK M&amp;M PA46-NY-09199-07-or later revisions of the above listed documents approved by EASA.</td>
<td>02/02/2018</td>
<td>Active</td>
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<td>BE AEROSPACE, Inc.</td>
<td>REV. 2</td>
<td>FAA STC 737-300-400-500</td>
<td>Validation of FAA STC No. ST03345AT issued 27 October 2006 Installation of Elementary Surveillance System (ELS) in the Boeing 737-300-400-500 series aircraft. Installation of ELS capable Mode S transponders and ELS capable ATC/TCAS Flight Instrument Control Panel.</td>
<td>Prior to installation of this design change it must be determined that the installation and/ or repair will introduce no adverse effect upon the airworthiness of the associated product.</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10038431</td>
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<td>HONEYWELL INTERNATIONAL INC.</td>
<td>REV. 1</td>
<td>FAA STC SR09583RC</td>
<td>Installation of the Sky Connect Forte Iridium Satellite Phone and (or) automated reporting system (FAA STC SR09583RC)</td>
<td>Prior to installation of this design change it must be determined that the installation and/ or repair will introduce no adverse effect upon the airworthiness of the associated product.</td>
<td>04/02/2015</td>
<td>Active</td>
<td>10038432</td>
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<td>ARMSTRONG AEROSPACE Inc.</td>
<td>REV. 2</td>
<td>FAA STC ST02947CH</td>
<td>Validation of FAA STC ST02947CH installation and activation of a Panasonic Global Communications Suite(GCS) system in accordance with Armstrong Aerospace Master Data List DC104-0003-00, Revision fl. dated 17 December 2012 Revision 1 of this STC has been issued in order to add the installation of the 20/30 angle and two axis Automatic Flight Guidance Systems. Installation Model 73-743 (28V System) according to bulletin N. 843.</td>
<td>Master Data List, Reference DC104-0943-00, Revision M, dated 17 December 2012. Instructions for Continued Airworthiness Supplement DC113-0943-00, Revision E, dated 23 July 2012. Weight and Balance supplement DC304-0943-00, Revision C, dated 16 June 2012. May 2012 Airplane Flight Manual Supplement DC303-0943-00, Revision A.</td>
<td>21/02/2013</td>
<td>Active</td>
<td>10038436</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>MITSUBISHI HEAVY INDUSTRIES AMERICAINC.</td>
<td>REV. 1</td>
<td>FAA STC  SA01545E</td>
<td>3-MU-2B-36, -36A, -40, -40A, -60, -60A, -60B, -60C, -60D, -60E, -60F, -60G</td>
<td>Fabrication of Stretched Acrylic Windows for Mitsubishi MU-2B series aircraft/Version 1 installs a stretched acrylic window on the MU-2B series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/04/2020</td>
<td>Active</td>
<td>10038450</td>
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<tr>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES,A.R.L.</td>
<td>REV. 1</td>
<td>FAA STC  SA01546E</td>
<td>3-MU-2B-30, -30A, -30B, -30C, -30D, -30E, -30F, -30G, -30H, -30I, -30J</td>
<td>Installation of Garmin GNS 400 series system, GA 35 GPS antenna, GMX 200multifunction display, GMA 347 audio box, SL30 NAV/COM system, GTX 338transponder, M&amp;I Continent MD-41 annunciator, Collins ALT 50/90/100/200/500/1000 and Indigo One Tracking system, according to SEAM STC No. 3622/11</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Flight Manual Supplement &quot;GNS 400 SERIES-B&quot; ref. SEAM- AFM-GNS, Revision 2 dated 26/07/2011: STC Data List ref. SEAM-SDL- BN2T, Revision 3 dated 27/07/2011: or later revisions of the above listed documents approved by EASA.</td>
<td>17/02/2012</td>
<td>Active</td>
<td>10038454</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>FAA STC  SA01547E</td>
<td>ATP</td>
<td>Installation of a Five Panel Universal EFIS EFIS For APT Aircraft with PNY28 and PNY28A-EngineRev. 1: Applicability/standard to include both PNY28 and PNY28A.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s)-CRI F-01 Protection from the Effects of HFIR [s]=[s] Issue 2 dated 13/February 2012-CRI H-01 Instructions for Continued Airworthiness (CA) on Electric...</td>
<td>-Scandinavian Avionics&lt;s&gt;[&lt;][&gt;] Approved Aircraft Flight Manual Supplement:&lt;s&gt;[&lt;][&gt;] 2051192-AM A Issue 7 dated 14 August 2017- Scandinavian Avionics&lt;s&gt;[&lt;][&gt;] Approved Aircraft Flight Manual Supplement:&lt;s&gt;[&lt;][&gt;] 2051192-AM B Issue 4 dated 14 August 2017- Scandinavian Avionics&lt;s&gt;[&lt;][&gt;] Approved Aircraft Flight Man...</td>
<td>15/08/2017</td>
<td>Active</td>
<td>10038456</td>
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<tr>
<td>DECA AVIATION ENGINEERING Ltd./d/b/a PAL AEROSPACE</td>
<td>NO.</td>
<td>FAA STC  SA01548E</td>
<td>CL-600-2B19</td>
<td>Indium Antenna Installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Transport Canada STC No. O-LSA09-344D Issue 1 dated 19 November 2008: DECA Aviation Engineering Drawing No. 5002572, Revision A or later approved revisions: DECA Aviation Engineering Instructions for Continued Airworthiness CA09215, Revision 1 or later approved revisions.</td>
<td>Restricted to ASN 8592.</td>
<td>17/02/2012</td>
<td>Active</td>
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<td>MYSTERE FALCON 20</td>
<td>REV. 1</td>
<td>FAA STC  SA01549E</td>
<td>Enhanced Mode &quot;S&quot; compliance upgrade applicable to DassaultAviation/MBB aircraft.</td>
<td>Enhanced Mode &quot;S&quot; compliance upgrade applicable to DassaultAviation/MBB aircraft.</td>
<td>The Certification Basis for the CB) for the original product remains applicable to this certificate/approval.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/01/2015</td>
<td>Active</td>
<td>10038465</td>
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<tr>
<td>LUFTHANDS TECHNIK AG</td>
<td>A340-311 / -312 / -313</td>
<td>FAA STC  SA01550E</td>
<td>A340-300 GPSSU ReplacementThe STC consists in replacing the current Global Positioning SystemSensor Unit (GPSSU)</td>
<td>A340-300 GPSSU ReplacementThe STC consists in replacing the current Global Positioning System Sensor Unit (GPSSU) by a new standard able to transmit GPS quality information.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>None.</td>
<td>21/02/2012</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
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<td>Model</td>
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<td>10038463</td>
<td>DASSAULT FALCON SERVICE</td>
<td>REV. 2</td>
<td>F900EX EASY</td>
<td>F900EX EASY</td>
<td>Major Change to EASA STC 10038490 This Major Change consists in removing the former limitation (only Version 1 F900EX EASY B-Stack 3+) was approved</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval</td>
<td>DFS Change Approval Sheet DFS-9986-CAS Issue B dated 05/04/2012</td>
<td>As per DFS Change Approval Sheet DFS-9986-CAS Issue B dated 05/04/2012</td>
<td>08/05/2012</td>
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<td>10038491</td>
<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td>REV. 1</td>
<td>AS335F2, AS355N, AS355NP, AS355F1</td>
<td>Installation of video transmission equipment ref. PP-014 Rev. 01: New cabin rack U12 (PN PP-10-4405) installation added.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated noise and emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List PP-014-MLD-01, issue 1, dated 02/04/2020 Flight Manual Supplement PP-014-AM-01, issue 1, dated 02/04/2020</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/04/2020</td>
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<td>10038498</td>
<td>ALPINE AVIATION, L.L.C.</td>
<td>FAA STC SAD0469NY</td>
<td>2-33, 35, 36 1-436</td>
<td>Installation of Main Landing Gear Auxiliary Landing Lights in accordance with FAA STC SAD0469NY Installation of Main landing Gear Auxiliary Landing Lights in accordance with Alpine Aviation, Inc. Installation Manual, Report No. 014, dated June 14, 1996.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated noise and emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Alpine Aviation, Inc. Installation Manual, Report No. 014, dated June 14, 1996</td>
<td>FAA approved</td>
<td>21/02/2012</td>
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<td>10038605</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>FAA STC ST02185SE</td>
<td>B747-400</td>
<td>Installation of a JMS RAHE In-Flight Entertainment System into the economy class passenger seats of a Boeing 747-400 aircraft in accordance with NW Aeronautics Technologies Inc. Master Drawing List N03314-D10, Rev. J dated December 27, 2011 or later approved revisions. Validation of FAA STC ST02185S...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Northwest Aerospace Technologies Inc. Master Drawing List N03314-D10, Rev. J dated December 27&lt;+,&lt;&gt;,+&lt; 2011 and documents referenced in this MDL or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2000/40/EC (or subsequent revisions off this decision).</td>
<td>None.</td>
<td>22/02/2012</td>
<td>Active</td>
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<td>10038512</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 1</td>
<td>CESSNA 425</td>
<td>Installation of 5-blade MTV-27-1-E-C-F-R (FVC7FR-A10-500 propellers, Cessna 425 aircraft and models modified by EASA STC EASA.M.03944 (FAA/9085/855Y) to install PFA-135 (A) engine).</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: FAR 23.1529 Amendment 287 This certificate/approval involves a change to the requirements for environmental protection and/or a change to...</td>
<td>Airplane Flight Manual Supplement E-2105, Revision 0, 16.01.2012</td>
<td>FAA approved</td>
<td>02/03/2012</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>STC No</th>
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<td>10038517</td>
<td>FIELD AVIATION EAST, LTD.</td>
<td>REV. 1</td>
<td>TCCA STC- SANS-8</td>
<td>DHC-8-202</td>
<td>Maximum Gross Takeoff Weight Increase to 37,300 lbs (c/w MZFW Increase and Ground Spoiler Options)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation must be in accordance with FAEL Master Drawings and Data List designated DCL1124. Revision L Flight Operations must be in accordance with FAEL Flight Manual/Supplements designated FMS 20083, Revision C and FMS 20187. Revision A Instructions for Continuing Airworthiness Maintenance must be p...</td>
<td>09/02/2015</td>
<td>Active</td>
<td>10038517</td>
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<tr>
<td>10038518</td>
<td>ING. MIROSLAV RUSINAK S.R.O.</td>
<td></td>
<td>BELL 427</td>
<td>Support unit for medical devices</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>07/10/01/03 initial release - Flight Manual Supplement - later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>22/02/2012</td>
<td>Active</td>
<td>10038518</td>
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<td>10038524</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td></td>
<td>FAA STC ST0394AT-D</td>
<td>GV-SP</td>
<td>Installation of an Audio Video Distribution System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List GC515128003, Revision C, dated June 23, 2010 or later revisions of the above listed documents approved by EASA.</td>
<td>23/02/2012</td>
<td>Active</td>
<td>10038524</td>
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<td>10038533</td>
<td>FIELD AVIATION EAST, LTD.</td>
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<td>TCCA STC C- LSA11-177/0</td>
<td>DHC-8-202</td>
<td>Cabin Drop Down Oxygen System Removal and Cabin Interior Refurbishment</td>
<td>FAR 25 as per EASA M.A.191, Issue 5 (incl. EWIS)</td>
<td>FAR 25 as per EASA M.A.191, Issue 5 (incl. EWIS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation must be in accordance with TCCA approved FieldAviationMaster Drawing and Data List 25470000, Revision A. Maintenance must be in accordance with TCCA accepted FieldAviationMaintenance Manual Supplement 25471200, Revision B. and/or later revisions of the above listed documents approved by EASA.</td>
<td>23/02/2012</td>
<td>Active</td>
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<td>10038558</td>
<td>BRISTOW HELICOPTERS Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>S-76C</td>
<td>Modification of Rockwell Collins TCAS-4000 TCAS II update to version 7.1 software.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>5765 Flight Manual Supplement No: 19 revision 1, 5765 Flight Manual Supplement No: 23 initial issue or later revisions of the above listed documents approved by EASA. BHEL/ S76,4552 initial issue dated 29 March 2012.</td>
<td>23/02/2012</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 5</td>
<td>SR22</td>
<td>Installation of 4-blade MTV-14-D115-30 ( SR22)</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/ approval. This certificate/approval involves a change with an impact on the noise database and/ or emission levels. The environmental protection requirements are amended as follows: Certificate Basis for No...</td>
<td>MT-Propeller Master Documentation List No. E-254/10 To be maintained i.e.w., MT-Propeller/ICA Doc. No. E-2123 To be operated in accordance with AFM-S/Doc. No. E-2124 for the SR22/SE-2125 for the SR22 or E-4034 for SR22 with RPM reduction to 2500RPM</td>
<td>According to limitation section of AFM-S</td>
<td>21/11/2022</td>
<td>Active</td>
<td>10038596</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>777-200</td>
<td>Introduction of Tailwind 550 Antenna With the revision 2 dfis (STC) the previous life limit of 500 Flight Cycles is removed.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Supplemental Type Definition Document, Lufthansa Technik document no. STD053-132 1M6 Issue 1, Revision 2, dated 10 January 2014 - Aircraft Flight Manual Supplement, Lufthansa Technik document No. A-00-11/177-AFM Issue 2, Revision 0, dated 21 March 2012 - Instructions for Continued Airworthiness, Lufhan...</td>
<td>This modification is limited to MSN 29953. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/05/2014</td>
<td>Active</td>
<td>10038569</td>
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<td>SII DEUTSCHLAND GMBH</td>
<td>REV. 1</td>
<td>777-200</td>
<td>Installation of CAIR (TM) system</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval-CR-I CA on EWIS</td>
<td>Master Document List MDA-217-00001-000A Iss. A dtd. 22 Feb 2012-ICA per ANM Supplement AMS-217-00001-000A Iss. A dtd. 16 Feb 2012 or later revisions of the above listed documents approved by EASA</td>
<td>This modification is limited to MSN 29953.</td>
<td>23/11/2012</td>
<td>Active</td>
<td>10038599</td>
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<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA</td>
<td>3. PA-22</td>
<td>Installation of vortex generators on the wings, horizontal and vertical surfaces (all models), and increase in maximum zero fuel weight/PA-34-200T and M20 03 only.</td>
<td>Design Non-significant change. The Certification Basis of the original product remains applicable to this change. Environmental Protection: The certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with FAA approved Micro Aerodynamics, Inc. Drawing Package MA2212, Revision A, dated 14-Aug-2006 and FAA approved Installation Manual MA2213, Revision A, dated 06-Oct-2006 Micro Aerodynamics PN60-6FM Supplement No. MA2215, Revision R1, dated 25-Oct-2006, for PA-34-200T and PZL...</td>
<td>If more than five vortex generators are damaged or missing, the aircraft is no longer airworthy.</td>
<td>27/02/2012</td>
<td>Active</td>
<td>10038573</td>
<td></td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>A330-200, -300</td>
<td>A330 Tape Video Reproducer Unit Installation The change consists in replacement of each MA5 Video Tape Player (VTP) by new digital videoreproducer units manufactured by Goodrich installed in the air traffic control Video Center.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval A330 EASA TCDS A. 004 The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>Certification Plan 4311-01-A-2101-R02 dated 24th February 2012 Master Data List 4311-01-A-2103-R01 dated 27th February 2012 Instructions for Continued Airworthiness 4311-01-A-2211-R00 dated 17th January 2012 Aircraft Configuration Index Document (ACID) 4311-01-A-2213 or later revisions of the above...</td>
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<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>BOMBARDIER AEROSPACE</td>
<td>10038657</td>
<td>1</td>
<td>1</td>
<td>LTS101-850B-2</td>
<td>Galaxy Modification for Microwave Overheat for the original product remains applicable to this certificate/approval.</td>
<td>TCCA STC No. SA12-12, Issue 1 dated 23 January 2013. ACI-NAE Ltd. Modification Data Summary Number H372000 Revision N/C date 23 January 2012 or later approved.</td>
<td>Instructions for Continued Airworthiness, ref. to Bombardier document BA63058004</td>
<td>Revision N/C or later approved revision or later revision.</td>
<td>29/02/2012</td>
<td>Active</td>
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<td>GULFSTREAM AEROSPACE</td>
<td>10038629</td>
<td>2</td>
<td>3</td>
<td>FAA STC STD4251AT-D</td>
<td>Installation of an Executive Completion - Multiple STC</td>
<td>This Executive Completion, Multiple STC, (FAA ref S704251AT-D), does not approve the entire airplane interior installation and has the following limitations as defined in document GC522043046. <em>Guidance for Follow-On Installation, Executive Completion Multiple STC</em>.</td>
<td></td>
<td>Complete airplane interior require...</td>
<td>23/02/2012</td>
<td>Active</td>
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<td>(UK) Ltd.</td>
<td>1003830</td>
<td>1</td>
<td>1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>Installation of new cabin interior as defined in HAACP075</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td></td>
<td>None</td>
<td>10/07/2012</td>
<td>Active</td>
<td>1003830</td>
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<td>AEROSPACE DESIGN AND</td>
<td>1003846</td>
<td>1</td>
<td>1</td>
<td>MBB-BK 117 B-2</td>
<td>Installation of LTS101-850B-2 Engines into MBB-BK117 B-2 Helicopter</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- Master Data List JN216 ADD17, Issue 2, Rev 2 dated 20/02/2012 - ICA JN216 ADD54-20 (1), Issue 1, Rev 3 dated 06/12/2011 - RFMS JN216 ADD 29 (1) Issue 1, Rev 6, dated 01/03/2012.</td>
<td>- Installation of LTS101-850B-2 Engines into MBB-BK117B-2, RFMS JN216 ADD 29(2) Issue 1, Rev 6, dated 01/03/2012.</td>
<td>None</td>
<td>05/03/2012</td>
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<td>T/A AERODAC</td>
<td>1003856</td>
<td>2</td>
<td>1</td>
<td>FAA STC SR02138SE</td>
<td>Installation of Dart D139-799 Maintenance kit. Note: STC re-issued at Rev 2 for addition of the AB139 model.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Dart Master Document List MOL-D139-799, Revision C, dated June 6, 2010.Dart Installation Instructions IN-D139-799, Revision D, dated June 10, 2011.Dart Instructions for Continued Airworthiness ICA-D139-799, Revision 3, dated May 9, 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>22/11/2017</td>
<td>Active</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>BOMBARDIER INC.</td>
<td>REV. 2</td>
<td>TCCA STC: SA12-24</td>
<td>BD-700-1A10, BD-700-1A11</td>
<td>Installation of Class 6 Electronics Flight Bag (EFB) System: Global 5000 and/or Global 5000 aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Transport Canada STC No. SA12-24 Issue 3 dated 15 August 2012; Bombardier Inc. Modification Data Summary Number RAZ- BM700-003; Revision A, approved 02 March 2012; Bombardier Aerospace Flight Manual Supplement D- FN46054001 Revision B for Continued Airworthiness refer to Bombardier document...</td>
<td>None.</td>
<td>09/15/2012</td>
<td>Active</td>
<td>10038659</td>
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<tr>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>REV. 2</td>
<td>TCCA STC: O-LSA11-232/D</td>
<td>DHC-8-103</td>
<td>FAR 25 as per EASA A.M.191, issue 5 (incl. ENS)</td>
<td>Installation must be accomplished in accordance with TCCA approved FAAST/Aviation Company Inc. Master Drawing and Data List designated 25550000, Rev. 0. Flight operations must be conducted in accordance with FAA Flight Manual Supplement designated FMS 25550000 Rev. 0. Instructions for Continuing Airworthiness refer to Aviation Company Inc. Master Drawing and Data List designated 25550000, Rev. 0.</td>
<td>This modification is limited to Serial number 210 with EASA STC No. MOT.21NE2.130 installed.</td>
<td></td>
<td>06/03/2012</td>
<td>Active</td>
<td>10038661</td>
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<tr>
<td>AVIATION PARTNERS, INC.</td>
<td>REV. 2</td>
<td>FAA STC STO0188SE</td>
<td>MF900, F900EX</td>
<td>F900 aircraft fitted with wingslet per EASA STC 10038667.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Certification Flight Test Plan rev. A dated August 14, 2012 and referenced AFPS-0601 Code 9001 Revision 1 April 15, 2013 Master Drawing List Revision D, dated December 6, 2011. These drawings are on the above listed documents approved by EASA.</td>
<td>Applicable to all Dassault Aviation MF 900 and F900EX aircraft fitted with EASA STC 10038667. For modification of this modification it must be determined that there is no adverse effect, interrelationship between this modification and any other previous/installed modification and/or repair will introduce no adverse effect, etc.</td>
<td></td>
<td>24/04/2013</td>
<td>Active</td>
<td>10038665</td>
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<tr>
<td>AVIDYNE CORPORATION</td>
<td>FAA STC SA030180</td>
<td>PA46-350P, PA-46R-350T</td>
<td>Autopilot Series / DFC 90 Autopilot P/N 700-00170-002 DFC100 Autopilot P/N 700-00170-003 as replacement for the Si-Tec System 55X Autopilot P/N192-6).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Operation must be in accordance with Aviation Corporation Aircraft Flight Manual Supplement (AFMS) as follows: For DFC90: Document 600-00285-000, Rev 06, FAA approved on November 17, 2011. For DFC100: Document 600-00288-000, FAA approved on November 17, 2011.</td>
<td>Aircraft must be configured with an existing 0-TEC 55X or 55R Autopilot and an existing Avidyne Flight Display System as follows in order to accomplish this STC: For DFC90: Avidyne Primary Flight Display, p/n 700-00060( ), loaded with release 8.0.4 software (or later approved release) with...</td>
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<td>06/02/2012</td>
<td>Active</td>
<td>10038669</td>
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<td>EASA STC No.</td>
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<td>Foreign/NAA STC ref.</td>
<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10038671</td>
<td>SAFRAN ELECTRONICS &amp; DEFENSE BUSINE SS UNIT - HELICOPTERS DATAMANAGEMENT NT SOLUTIONS &amp; SERVICES</td>
<td>REV. 3</td>
<td>EC 225 LP</td>
<td>Installation of the Helicom V2 NUM embedded Equipment:Revision 1: Add the possibility of installing Helicom V2+ NUM with Cargorelease information (cyclic stick and collective lever) Revision 2: Adding Air Intraes data acquisition and correcting typo oCMa and RNFMS references Revision 3: STC holder n...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Nk-EC225 HV2+ NUM-RNFMS-E-A-03 Rev.00 dated 20 January 2016 – Rotorcraft Flight Manual Supplement: N8-EC225 HV2+ NUM-E-A-03 Rev.01 dated 20 January 2016 – Instructions for Continued airworthiness– N1. EC225 HV2+ NUM-E-A-02 dated 29/11/2017 – List of Document/0 later revisions of the ab...</td>
<td>The approval holder shall fulfill the obligations of Part 21, Paragraph 21.105.Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/02/2020</td>
<td>Active</td>
<td>10038671</td>
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<td>10038677</td>
<td>DSA A.S.</td>
<td>REV. 1</td>
<td>F172P F172N F172M</td>
<td>Avionics upgrade - installation of GMA 340 (audio), SL 40 (VHF COMM), SL 30 (VHF NAV/COMM), HD 320-328 (CDI indicator), GTX 328 (Mode S XPNDR), CI 102C (MIKtremet), Kannad 400-AP (Compact (ELT), AV 200 (ELT antenna), CI 105/GPXNDR antenna).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any in...</td>
<td>- Cassina F172 M, N, P Flight Manual Supplement No: S7 LDA 0211/ C127GMA 340 Audio Panel, issued: May 30, 2011 - Cassina F172 M, N, P Flight Manual Supplement No: S2 LDA 0211/EC225SL30 VHF COMM Transceiver NAV Receiver, issued: May 30, 2011 - Cassina F172 M, N, P Flight Manual Supplement No: S3 LDA 02...</td>
<td>To prior installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/05/2020</td>
<td>Active</td>
<td>10038677</td>
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<td>10038678</td>
<td>SABENA TECHNICS BOC</td>
<td></td>
<td>B777-200</td>
<td>0527-01: eXPhone System Installation on B777-200 aircraft.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval: CS 25 Amdt 4 – CRI F01 GSM System (ref AEC STC 0333-01) – CRI F02 Hardware (ref AEC STC 0333-01) – CRI F03 Software (ref AEC STC 0333-01). MDL 0527-01-2103-R00 dated 6 March 2012 – AFM Supplement 0527-01-A-2301-R00 dated 17 Feb 2012 - Instructions for Continued Airworthiness 0527-01-A-2311-R00 dated 10 Nov 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>Per AFM Supplement 0527-01-A-2301 this modification is limited to Boeing 777-200.</td>
<td>07/03/2012</td>
<td>Active</td>
<td>10038678</td>
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<td>10038683</td>
<td>AIM AVIATION (HENSHALLS) LTD</td>
<td></td>
<td>A321</td>
<td>Installation of Galley 2A Part No. 321B92DA004 and Galley 3A Part No321B8M5AG004 into an A321-200 to be operated by BMI.</td>
<td>CS 25 Amendment 11The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Change Record HEN 1207, Issue 1</td>
<td></td>
<td>None</td>
<td>07/03/2012</td>
<td>Active</td>
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<td>10038702</td>
<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINES A.R.L</td>
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<td>DR-400/180</td>
<td>STC 3725/11 rev. 0 dated 04/01/2012 – Amelioration equipments radioavion-RNAV</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification notice and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplement Manu de Vol SMV 3725-11 rev. 0 du 04/01/2012 or later revisions of the above listed documents approved by EASA.</td>
<td>S/N 2105</td>
<td>09/03/2012</td>
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<td>10038717</td>
<td>(EASA.A.S.00327) BUSINESS AND COMPUTER AIRCRAFT</td>
<td>REV. 1</td>
<td>B190DD</td>
<td>Installation of Two Transponders mode S RC2.452</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval: FAR 23 Amdt 23-62The certification notice and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>File ref - BCA 60 Rev 1 dated 11/01/2012Installation ManualRC2.452 No. A 19-3851-001</td>
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<td>ALTITUDE AEROSPACE INC.</td>
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<td>TCCA STC SA11-43</td>
<td>DHC-8-201-202</td>
<td>Installation of an Oven and a Coffee Brewer in the Galley.TCCA STC SA11-43, Issue 1</td>
<td>FAR 25-25 as per EASA BM A.191, Issue 5 (incl. SC H31 ICA on (EWIS).The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product area...</td>
<td>Installation must be in accordance with Altitude Aerospace Inc. Modification Summary M83200044, Revision E, dated 25 April 2011.Maintenance must be in accordance with Altitude Aerospace Inc. Instructions for Continued Airworthiness ICA 260044, Rev. A, dated 17February 2010 or later revisions of the abo...</td>
<td>Galley PN H82530009 only</td>
<td>09/03/2012</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 2</td>
<td>FAA STC 1063</td>
<td>Installation of MTV-14-D106-360</td>
<td>For MTV-14-D106-360 to be operated in accordance with MT-Propeller FAA STC Document No. E-4004Rev. 0 to be installed i.e. MT-Propeller Master Documentation List No. E-2018 Rev. 0 to be maintained i.a.w. MT-Propeller ICA Doc. No. E.</td>
<td>04/10/2022</td>
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<td>JET AVIATION AG</td>
<td>REV. 3</td>
<td>FAA STC 1038621</td>
<td>Installation of an Airplane Cabin Interior.</td>
<td>According to limitation section of AFM-S. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>11/02/2017</td>
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| BLR AEROSPACE, L.L.C. | REV. 2 | FAA STC 1038758 | Installation of a BLR Aerospace Ultimate Performance Package (UUP) | 1. The approved EASA Configurations and the compatible EASA export options are defined in GCS22040616, "EASA Validated Configuration Lists"<(...),<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>,<>
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<th>Rev. level</th>
<th>Foreign/NAA Code</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>10038764</td>
<td>BOMBARDIER INC.</td>
<td>TC2A STC C-LISA12-008D</td>
<td>BD-700-1A11</td>
<td>Installation of Complete Custom Aircraft Interior - ASN 9401</td>
<td>EASA M.4 A.2097</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Restricted to ASN 9401</td>
<td>12/03/2012</td>
<td>Active</td>
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<td>10038766</td>
<td>SAFRAN CABIN INC.</td>
<td>FAA STC NO. ST00415LA</td>
<td>B767-300</td>
<td>Interior Installation and Modification to include Center Bin, Cabinceilings and Outboard Stowage Bins.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. ST02415LA dated 23 February 2012 or later approved revision; instructions for Continued Airworthiness (CAA), C(&gt;&amp;&lt;)&gt;D Zodiac Master Document No. MDOL 2027423_20120001 and Gomolzig Installation Instructions Doc. No. 8220423_rev0_20120201 or later approved revision.</td>
<td>None.</td>
<td>12/03/2012</td>
<td>Active</td>
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<tr>
<td>10038775</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 3</td>
<td>PA-25-235, PA-25-260</td>
<td>Low Noise system PA-25-235-605600 Modification 2027423 applicable to PA-25-235 only.</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change with an impact on the noise/database and/or emission levels. The environmental protection requirements are amended as follows: CS-36 Amdt 2Thres.</td>
<td>Low Noise system PA-25-235-605600 Modification 2027423 applicable to PA-25-235 only.</td>
<td>None.</td>
<td>12/03/2012</td>
<td>Active</td>
<td>10038775</td>
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<td>10038783</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE AUTIQUE (SÉGA)</td>
<td>REV. 1</td>
<td>2. AS355F, AS355F1, AS355F2</td>
<td>Front Lower Cowling Modification (Reference H-01-11-Capot) Separation of cowling in two halves reassembled by fasteners. Note: this STC is also applicable to front lower cowlings modified after H-01-11-Capot.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>According limitation section of AFM-S. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/12/2023</td>
<td>Active</td>
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<td>10038786</td>
<td>SHERDELL ENGINEERING Ltd.</td>
<td>8. VENTUS BT, 9. CIRRUS</td>
<td>Cockpit Ventilation Modification</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>According limitation section of AFM-S. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>None.</td>
<td>14/03/2012</td>
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<td>HAWKER BEECHCRAFT LTD.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>To introduce increased maximum landing weight to 16,500lbs for the KingAir B300GCER.</td>
<td>Installed in accordance with Hawker Beechcraft Limited Major Change ref SC040409a at issue 1. Operation in accordance with Hawker Beechcraft Limited Aircraft FlightManual Supplement 2, issue 1 to AFM ref 130-50031-245, Maintaine d in accordance with instructions for Continued Airworthiness Document AIRCRAFT.</td>
<td>None.</td>
<td>14/02/2012</td>
<td>Active</td>
<td>10038800</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>Installation of Seat Rail Mounted Fuel Tank.</td>
<td>The Certification Basis for the original product remains applicable to this certificate.</td>
<td>None.</td>
<td>20/03/2012</td>
<td>Active</td>
<td>10038818</td>
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<td>VT MOBILE AEROSPACE ENGINEERING, Indio/ VT MAE</td>
<td>REV. 3</td>
<td>FAA ST03952AT</td>
<td>Modification of Passenger Aircraft to Cargo Passenger Configuration. Rev. 3: Modification of 8 Pallet 80 Passenger Cargo Passenger (Comb) to a fourteen pallet (14P) freighter configuration.</td>
<td>The Certification Basis for the original product remains applicable to this certificate.</td>
<td>None.</td>
<td>03/03/2020</td>
<td>Active</td>
<td>10038822</td>
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<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td>A319-115</td>
<td>Airbus Corporate Jet Centre Modification CJ0950 - VIP Interior - (Metropol) Revision due to update of AFM-S and MODS</td>
<td>The Certification Basis for the original product remains applicable to this certificate.</td>
<td>None.</td>
<td>25/10/2013</td>
<td>Active</td>
<td>10038826</td>
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<td>BUCHER LEICHTBAU AG</td>
<td>REV. 1</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Small person restraint system P/N 1120040 for use on 160 New Generation/stretcher P/N 1120020-XXX.</td>
<td>This STC is limited to A319-115</td>
<td>20/02/2012</td>
<td>Active</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10038857</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERINGCO RPORATION AG (AIEC)</td>
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<td>CL-600-2B16</td>
<td>Indium Satcom Installation (ICG)</td>
<td>EASA.IM.A.023</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>20/03/2012</td>
<td>Active</td>
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<td>10038859</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERINGCO RPORATION AG (AIEC)</td>
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<td>CL-600-2B16</td>
<td>Terrain Awareness and Warning System (TAWS) Upgrade to SCN 12.0 (UASC)</td>
<td>EASA.IM.A.023</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>20/03/2012</td>
<td>Active</td>
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<td>10038860</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERINGCO RPORATION AG (AIEC)</td>
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<td>CL-600-2B16</td>
<td>CVR-120 replacement by CVR-120R (UASC)</td>
<td>EASA.IM.A.023</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>20/03/2012</td>
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<td>10038884</td>
<td>BOMBARDIER INC</td>
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<td>TCCA STC SA11-96</td>
<td>BD-700-1A10 Installation of a Frequency Converter Fuse</td>
<td>EASA.IM.A.009</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>20/03/2012</td>
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<td>10038866</td>
<td>VENTURA AEROSPACE, Inc.</td>
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<td>FAA STC STG0118-A</td>
<td>B767-200 Installation of 9g Forward Cargo Barrier on the Main Deck</td>
<td>EASA.IM.A.009</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. For areas for areas of change the Certification Basis is CS25 Amendment2 Requirements at amendment 2 are 25.301, 25.305, 25.307, 25.561, 25.571, 25...</td>
<td>None</td>
<td>20/03/2012</td>
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<td>STC Holder</td>
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<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STD 0305QAT</td>
<td>Installation of a Lead Acid Battery. The STC installs a Lead Acid battery in the existing high-voltage Cadmium battery.</td>
<td>Master Drawing Unit (MDU), ref: 5-0372, Revision D, dated 16 September 2011. Instruction for Continued Airworthiness, ref: 5-0446, Revision A, dated 13 September 2011. Aircraft Flight Manual Supplement, ref: 5-0374, dated 20 June 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>20/03/2012</td>
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<td>ANDRA INTERNATIONAL LLC</td>
<td>B767-200</td>
<td>Installation of Main Deck Cargo Loading System.</td>
<td>Anora International LLC Controlled Data List, DL55110, Revision B, dated 10/10/2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/044(0) or subsequent revisions of this decision.</td>
<td>None.</td>
<td>21/03/2012</td>
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<td>MD-CANADA MOD CENTER</td>
<td>CL-600-2B19</td>
<td>Installation of Main Deck Cargo Loading System.</td>
<td>Transport Canada STC No. SA10-16, Issue 1, dated 03 February 2010. Avionics Design Services Ltd Master Drawing List MDL09134, Revision N/C, dated 03 February 2010, or later revisions accepted by Transport Canada.</td>
<td>None.</td>
<td>21/03/2012</td>
<td>Active</td>
<td>10038889</td>
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<td>BOURNEMOUTH AVIATION CONSULTANTS LTD</td>
<td>CAA UK RESPONSIBILITY</td>
<td>OBSERVER MOVING MAP SYSTEM INSTALLATION The system is manufactured by Skyforce and consists of a 5&quot;Display/Control Head, a Processor Unit, and a Control Panel. The equipment is CAA approved under EASA Annex 6-10 Supplement No. 0542-01-2103-R00. The installation is identified and defined in the document &quot;Stove&amp;Heat...&quot;</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate. For areas of change the Certification Basis is C255 Amendment 10. Requirement at amendment 10 are 25.301, 25.303, 25.305, 25.307, 25.471, 25.561(...).</td>
<td>EASA MA.023 The Certification Basis for the original product.</td>
<td>None.</td>
<td>22/03/2012</td>
<td>Active</td>
<td>10038890</td>
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<td>BOURNEMOUTH AVIATION CONSULTANTS LTD</td>
<td>CAA UK RESPONSIBILITY</td>
<td>OXYGEN SYSTEM MODIFICATION Replacement (Supplementary) P/N 1270512-3 Oxygen cylinder with (now) &quot;British Oxygen Company (BOC)&quot; Medical Gases AF size cylinder. The installation is identified and defined in the &quot;Stove&amp;Heat&amp;Heat...&quot;</td>
<td>The Certification Basis for the original product. Remaining applicable to this certificate.</td>
<td>None.</td>
<td>22/03/2012</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD.</td>
<td>CA UK RESPONSIBIL ITY</td>
<td>A100E</td>
<td>Installation of Bendix/King RDR 2000 Weather Radar System (Bendix/King RDR 200 in CAA approved Ref. VC 0104B). The installation consists of an antenna-receiver-transmitter (ART) unit, the Configuration Module, and the Control Panel. The (ART) radar unit is installed in the nose compartment. The install...</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise/ emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>22/03/2012</td>
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<td>10038906</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD.</td>
<td>CA UK RESPONSIBIL ITY</td>
<td>A100E</td>
<td>Trimmie 2101 (10) Approach Plus GPS (CAA approved Ref. VCD0104B) The installation consists of a Jeppeesen data base slot card, a Receiver/Display Unit (RDU) interfaced to the EFIS via ARINC 429 and fitted on the top centre of the instrument panel. A GPS antenna is mounted on top of the tail fin. The install...</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise/emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>22/03/2012</td>
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<td>10038907</td>
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<td>BEECHCRAFT CORPORATION</td>
<td>REV. 2</td>
<td>FAA SA02130SE</td>
<td>KING AIR B200, B200C, B200GT AND B200GCT</td>
<td>Transfer EASA STC 10038906 Rev.1 Installation of S/LR Harboll Composite Propeller HC-EAN-3A/NG3028K.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certification/approval. Certification basis defined on FAA STC SA02130SE. This certification/approval involves a change to the requirements for environmental protection.</td>
<td>Installation in accordance with Master Drawing List (MDL) Document BLR-KA250-001, Revision A, dated May 3, 2011, and Installation Instr. Document BLR-KA250-004 Revision C, dated May 3, 2011. Operation in accordance with Airplane Flight Manual Supplement Document AFM-1135 Series No.7-1, Revision 1, dated ...</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/06/2014</td>
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<td>10038909</td>
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<td>MECAR AVIATION GROUP S.p.A.</td>
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<td>MBB-BK117 C-2</td>
<td>EC145 VIP MB Interiors Drawing No. 86EC10000001-001</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise/emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>None</td>
<td>23/03/2012</td>
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<td>AEROFDC WA LP</td>
<td>REV. 1</td>
<td>FAA STC SR02056SE</td>
<td>EC135 P2+, EC135 P2</td>
<td>Installation of an engine IFR system in accordance with the EC/2155/2004/EU Supplement EC/2155/2004/ENG-1S.6 and FDC/2155/2004/ENG-1S.6 and FDC/2155/2004/ENG-1S.6.</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise/emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>None</td>
<td>27/08/2012</td>
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<td>Limitations/Conditions</td>
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<td>THE NORDAM GROUP LLC</td>
<td>REV 1</td>
<td>FAA STC ST069346C</td>
<td>A319, A320, A321</td>
<td>Installation of C &amp; X Band Fuselage Nose Radome Shell</td>
<td>In accordance with the EASA TCDs. The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable with certificate approval SC-616. Enhanced Airworthiness Programme for Aeroplane Systems - ICA on EWIS as included in TCDs EASA A-064...</td>
<td>FAA STC ST069346C Master Data List, ref. doc. MDO-190-2400-2, Rev. E, dated 29th July 2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions off this decision)</td>
<td>None</td>
<td>27/03/2012</td>
<td>Active</td>
<td>10038928</td>
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<td>FIELD AVIATION COMPANY Inc.</td>
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<td>DHC-8-301/-311/-314/-315</td>
<td>DHC-8-302/-303/-304/-201/-202</td>
<td>Auxiliary Wing Fuel Tank Installation TCCA STC SA11-64, Issue 2</td>
<td>FARR 25 as per EASA M.A-191, Issue 3 (incl. SC-H-610). The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable with certificate approval Special Condition CRI E-01 (Fuel Tank Safety). The requirements for environmental protection...</td>
<td>DHC-8-100 and 200 Series Installation must be performed in accordance with FACI Master Drawing and Data List designated 2653000, Revision B. Flight Operations must be performed in accordance with FACI Flight Manual Supplement FMS 2018, Revision C. Instructions for Continuing Airworthiness Maintenance must...</td>
<td>None</td>
<td>23/03/2012</td>
<td>Active</td>
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<tr>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td></td>
<td>SR22, SR20</td>
<td></td>
<td>Installation of PFD Aspen EFD1000 PRO.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable with certificate approval CS-23, Amendment 2 (not required by 21A.101). The requirements for environmental protection and the associated ECL of noise and/or emissions levels of the original product must be unchanged and remain applicable to this certificate/approval Flight Manual Supplement ASR-2011-060-FLS-05-00 Rev. 00. or later revisions of the above listed documents approved by EASA. Instructions for...</td>
<td>No. Flight Manual Supplement ASR-2011-060-FLS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for...</td>
<td>None</td>
<td>23/03/2012</td>
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<td>DELTA ENGINEERING CORPORATION</td>
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<td>A330-223/-243</td>
<td>A330-201/-202/-203</td>
<td>PANASONIC GLOBAL COMMUNICATION SYSTEM. System installation, This STC installs Panasonic Global Communication System (SCS) and activates the wireless portion. The Certification Basis for the original product must be unchanged and remain applicable to this certificate/approval Master Data List, ref. 0154-07345-2269, revision F, dated 09 February 2012. Aircraft Flight Manual Supplement, ref. 0654-07840-2269, dated 03 October 2011 (Aircraft Configuration 1 Aircraft Flight Manual Supplement, ref. 0654-07841-2269, dated 04 January 2012 (Aircraft Configuration 2 &amp; 3) Instructions...</td>
<td>No. Flight Manual Supplement ASR-2011-060-FLS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA. Instructions for...</td>
<td>None</td>
<td>23/03/2012</td>
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<td>AEROTEC &amp; CONCEPT</td>
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<td>PA-23-250</td>
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<td>Installation of Aspen EFD 1000 Pro electronic flight instrument system.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable with certificate approval CS-23, amendment 1. The associated noise and emissions levels of the original product must be unchanged and remain applicable to this certificate/approval. Aerotec Flight Manual Supplement No. AE11143SMV-2 revision 2 dated 07/03/2012. Aerotec Flight Manual Supplement No. AE11143-2M version 1 dated 22/03/2012. Aerotec Flight Manual Supplement No. AE11143CA-2M version 0 dated 28/10/2012 or later revisions of the above listed documents approved by EASA. Initial installation of this design change must be performed by AEROTEC/Electrical personnel. This installation is approved by BRNAV operations with coupling TEC-55 autopilot The modified airplane must be equipped with an attitude indicator, an airspeed indicator, an attitude indicator, a magnetic...</td>
<td>Aerotec Flight Manual Supplement No. AE11143SMV-2 dated 07/03/2012. Aerotec Flight Manual Supplement No. AE11143-2M version 1 dated 22/03/2012. Aerotec Flight Manual Supplement No. AE11143CA-2M version 0 dated 28/10/2012 or later revisions of the above listed documents approved by EASA.</td>
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<td>Limitations/Conditions</td>
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<td>10038951</td>
<td>STC Holder</td>
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<td>208, 208B</td>
<td>Installation of Honeywell KFD840 Primary Flight Display.</td>
<td>The Certification Basis for the original product and the follow-on/additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>2011_026_C026_0506 Rev. 2 or later revisions of the above listed documents/approved by EASA. Installation Manual (IM) with drawings: 2011_026_C026_0500 Rev. 0. Instructions for Continuation/Airworthiness (ICA) with component configuration list 2011_02.</td>
<td>None.</td>
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<td>B767-200/-300</td>
<td>Installation of Aircraft Communications Addressing and Reporting System (ACARS) Mark II.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Drawing List 669-5007, revision M, 03/11/2005 AFM Supplement 758-0181-010 Revision A, 17/11/2005 or later revisions of the above listed documents/approved by EASA in accordance with EASA ED Decision 2004/44/CF (or subsequent revisions of this decision).</td>
<td>None.</td>
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<td>STC Holder</td>
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<td>DHC-6-100/-200/-300</td>
<td>Installation of Dual or Single Garmin GTN 650 or GTN 750 in DHC-6 series 100, 200, and 300 Twin Otter aircraft based on Engineering Data Summary E102910 &amp; E102911 in accordance with Transport Canada STC SA11-79, Issue 1.</td>
<td>The Certification Basis for the original product and the follow-on/additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
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<td>C-212-EE</td>
<td>TDM10-05 Rev. G titled &quot;Cabin Layout&quot;, dated 2012-03-30. &quot;New Cabin Layout for Maritime Surveillance System in CASA C212&quot;. Installation and Mounting of Maritime Surveillance System and change of Cabin Layout to accommodate a maximum of 5 crewmembers and 2 passengers in the Cabin. (Maximum number of passengers is 7.)</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval.</td>
<td></td>
<td>All equipment necessary for Maritime Surveillance Missions shall be installed or added with EASA DOA 21.108 Minor Design Change Approvals. Prior to installation of this change/repair it must be determined that the installation does not introduce any adverse effect on the airworthiness of the product.</td>
<td>12/09/2023</td>
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<td>EC135 T2(CPDS), EC135 T2+ (CPDS), EC135 T1(CDS), EC135 T1(PAS), EC135 P2(CPDS), EC135 P2+ (CPDS), EC135 P1(CDS), EC135 P1(PAS)</td>
<td>EMS Equip. PN 135-0004371-501 The equipment includes up to two Stretchers, two seats installed in the forward part of the cabin and slat Rear between the two seats. Rev. 1. Change of STC holder from &quot;Aerolite Air&quot; to &quot;Aerolite AG&quot;.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Flight Manual Supplement: FMS-135-000471-501-0 rev N/C dated 9.12.2011- Airworthiness Limitations: AORL-135-000471-501-0 rev N/C dated 9.12.2011 or later revisions of the above listed documents/approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the installation does not introduce any adverse effect on the airworthiness of the product.</td>
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<td>10038986</td>
<td>STC Holder</td>
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<td>BOEING 757-200</td>
<td>Installation of 21 Horse Stalls and a 10 Passenger Pallet.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>BAC Ltd Master Data List No. BAC-826ML05 Iss 1 dated 27 March 2012 or later revisions of the above listed documents/approved by EASA.</td>
<td>This EASA STC does not approve the installation of Horse Boxes. The following BAC STC Certifications must be installed as prerequisites for this approval: - Mod BAC-826MOD001 - Mod BAC-826MOD002 - Mod BAC-826MOD003 - Mod BAC-826MOD004 - Mod BAC-826MOD006</td>
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List of EASA Supplemental Type Certificates

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<td>Installation of main wheels and brake system P/N: JA-01 Revision 1: Update with new wheel P/N: RF-002(B) with brake EA-2065(s) which supersedes the P/N: JA-01 Revision 2: Update with new wheel P/N: RF-019 which supersedes the wheel P/N: RF-002(B) Revision 3: Update with new servicing instructions</td>
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<td>10039031</td>
<td>A330-300 Cabin reconfiguration</td>
<td>A330-321/-322/-323</td>
<td>The Certification Basis for the original product and the followon/diagnostic or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Master Drawing 1, Ref: BOF5034-D01, Revision E, dated 22 March 2012 Instruction for Continued Airworthiness Supplement. Ref: BOF5034-D15, Revision A, dated 12 March 2012.</td>
<td>None</td>
<td>30/03/2012</td>
<td>Active</td>
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<td>10039039</td>
<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td>EC635 T3</td>
<td>Installation of a Max-Viz EVS 1500 vision enhancement system</td>
<td>Rotorcraft Flight Manual Supplement PP-015-ADF-03 Master Data List PP-015-MDL-05 Certification Plan PP-015-CP-05 Instruction for Continued Airworthiness PP-015-IA-00 Wiring Diagrams PP-015-WD-03 Minor Change Definition PP-015- MCD4-00 Issue 0 dated 30 April 2018;ove listed documents approved/accepted.</td>
<td>Prior to installation of this design change it must be determined the thread interrelation between this design change and any other previously installed design change and if/ or repair will introduce aerochie effect upon the airworthiness of the product.</td>
<td>23/07/2018</td>
<td>Active</td>
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<td>10039040</td>
<td>E4C AEROSPACEx, Inc.</td>
<td>B767-300</td>
<td>Seat power, Seat Modifications and Seat Track installation</td>
<td>E4C master Data List 1993-000, revision D, 21/11/2007 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>None</td>
<td>15/05/2012</td>
<td>Active</td>
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<td>10039042</td>
<td>DART AEROSPACEx Ltd.</td>
<td>206 A/B/L/L1/L3/L4</td>
<td>Screw In Windshield Modification.</td>
<td>- Master Drawing List MCD4-00 Issue 1, dated 23 March 2012.</td>
<td>None</td>
<td>30/04/2012</td>
<td>Active</td>
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<td>10039051</td>
<td>AIRBUS S.A.S.</td>
<td>A321-200</td>
<td>Manufacturer’s Flight Manual Supp CJ1094, Issue 1 - Installation of Business Class Seating Area: “Dali 2 (Type C)”</td>
<td>- Modification Approval Sheet, ref: CJ1094, Issue 1, dated 2nd April 2012, Airplane Flight Manual APM Supplement, ref: FA-M-00-J7000, Issue 1 dated 1 April 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/05/CF (or subsequent revisions of this decision).</td>
<td>None</td>
<td>03/04/2012</td>
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<td>BLR AEROSPACE, L.L.C.</td>
<td>REV 3</td>
<td>FAA STC SR0166SS</td>
<td>BELL 204B, 205A1 BELL 212</td>
<td>Installation of Dual Tailloom Brakes and FastFin Revision 1 adds Model 204B and changes the reference to U.S.A STC RC 1163b to EASA STC 10040783. Revision 2 adds additional Bell 212 and 412 FastFin Revision 3 updates the STC holder's address.</td>
<td>The Certification Basis (CB) for the original product and the following/additional or alternative airworthiness requirements are applicable to both certificates/approval for components and areas affected by the change e.g., FastFin/SR0166SSS and STBS (STC SR0798SSS) in compliance with Commission Regu...</td>
<td>EASA approved Rotorcraft Flight Manual Supplement for Load Attachment Device BRD01 when installed in Bell Helicopters Model 205A-1 Series, Doc. No.: 349A-RMFS, Rev.: <em>None</em></td>
<td>22/05/2019</td>
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<td>RUIAO AEROSPACE SERVICES GmbH</td>
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<td>BELL 205A-1</td>
<td>Title: Load Attachment Device BRD01 on Bell 205A-1 Description: The BRD01 is a cabin mounted device for attaching loads on Bell helicopters. It is used with open sliding doors and bcan be mounted left- and right-hand simultaneously.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to both certificates/approval for components and areas affected by the change e.g., FastFin/SR0166SSS and STBS (STC SR0798SSS) in compliance with Commission Regu...</td>
<td>EASA approved Rotorcraft Flight Manual Supplement for Load Attachment Device BRD01 when installed in Bell Helicopters Model 205A-1 Series, Doc. No.: 349A-RMFS, Rev.: <em>None</em></td>
<td>04/04/2012</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
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<td>TAIFUN 17 E &amp; E II</td>
<td>Installation of a 2-blade MT-1-4K, 165-08 propeller on powered sailplane Taifun 17 E and Taifun 17 E II with Limbach L2400 EB1/LB or L2400 EI/DF engine</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to both certificates/approval for components and areas affected by the change e.g., FastFin/SR0166SSS and STBS (STC SR0798SSS) in compliance with Commission Regu...</td>
<td><em>None</em></td>
<td>10/04/2012</td>
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<td>KAMAN CORPORATION RCRAFT WHEEL AND BRAKE, LLC</td>
<td>FAA STC SA195G</td>
<td>2. B96, C90, E90, H90 (T-44A) 2. 65-400-4 (RU-21E, RU-21H) 2. 65-400-1 (UJ-21A) 2. 65-90, 65-A0 1. C90A, C90GT, C90GTI</td>
<td>EASA Validation of FAA STC SA195G Installation of Parker Hannifin Main Wheel and Brake Conversion Kit Parthumber 199-90 in accordance with Parker Hannifin Conversion Kit Installation Manual IM 199-90, Revision J, dated January 7, 2009, or later FAA approved revisions.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to both certificates/approval for components and areas affected by the change e.g., FastFin/SR0166SSS and STBS (STC SR0798SSS) in compliance with Commission Regu...</td>
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<td>10/04/2012</td>
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<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A321-213/-211-232 A321-211/-231-211-212 A320-216/-211-232-233 A320-111/-211-212-214/-215 A319-111/-120/-133 A319-111/-120/-113/-114/-115</td>
<td>In-Seat Power Supplies</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to both certificates/approval for components and areas affected by the change e.g., FastFin/SR0166SSS and STBS (STC SR0798SSS) in compliance with Commission Regu...</td>
<td>Modification TEN651 Issue 199-01 Manual Supplement TENA32003 Issue for later revisions of the above listed documents approved by EASA</td>
<td>12/04/2012</td>
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### List of EASA Supplemental Type Certificates

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<th>Rev. or level</th>
<th>Foreign/NAA STC ref.</th>
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<th>Description</th>
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<td>TATENHILL AVIATION Ltd.</td>
<td>CAA JT RESPONSIBILITY</td>
<td>PA-44-180 PA-44-180T PA-44-220T PA-44-220T</td>
<td>Approval of a Transponder System in accordance with AMC 20-2</td>
<td>CS23 Amendment 2The Certification Basis for the original product and the followning/additional or alternative airworthiness requirements are applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original...</td>
<td>Installation Manual M 00611 Revision A, dated 16 January 2013 Master Document List MDL.00611, Revision A, dated 29 March 2012</td>
<td>None</td>
<td>12/04/2012</td>
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<tr>
<td>PRATT &amp; WHITNEY ENGINE SERVICES, IN</td>
<td>FAA STC SA003080</td>
<td>C90GT, C90GT1</td>
<td>ADAS for Hawker Beechcraft Model C90GT, C90GT2 Installation of the Pratt &amp; Whitney Engine Services ADAS in accordance with Pratt &amp; Whitney Engine Services’ Installation Manual DIAS-6 0-045-5 Revision B, dated April 7, 2010, or later FAA approved revision</td>
<td>Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval. As established by the FAA STC SA003080B0, the requirements for...</td>
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<td>PRATT &amp; WHITNEY ENGINE SERVICES, IN</td>
<td>FAA STC SA002080</td>
<td>2.300 B200C B300 B300</td>
<td>ADAS for Hawker Beechcraft Model B200, B200C, 300, B300, B300C Installation of the Pratt &amp; Whitney Engine Services’ Master Document List No. DIAS-6-T-100-3, Revision D, dated May 13, 2010, or later FAA approved revision</td>
<td>The Certification Basis for the original product and the followning/additional or alternative airworthiness requirements are applicable to this certificate/approval. As established by FAA STC SA002080B0, the requirements for environmental protection and the associate...</td>
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<td>ATELIER CHABORD</td>
<td>REV. 2</td>
<td>CESNMA 172R 172S 172R 172S</td>
<td>Exhaust system and silencer installation (according to Chabord modification STC C172-12-01) Revision 2: Enhanced silent block/Alternative anti-corrosion treatment -Enhanced shape of central structural part in muffler</td>
<td>The Certification Basis for the original product as amended by the following/additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 23 Ard.3 § 607.611, 783, 807, 867, 1101, 1121, 1123, 1125 and 1529 The requirements for...</td>
<td>Technical Dossier ref: STC C172-12-01 Revision B dated 23 July 2020 Or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>05/08/2022</td>
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<td>RUAG AG</td>
<td>REV. 3</td>
<td>DHC-6 SERIES 300 DHC-6-300</td>
<td>Minor change to STC EASA A.S.00200 by adding the option to install a Honeywell MK VI incl. KMD-850 in a C172-12-01 installation of TAWS class A</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or...</td>
<td>-Master Document List, Doc. No. T6-000, Rev. 000, dated 15.05.2006- AFMS Doc. No. T6-500, Rev. 1, dated 15.05.2006 Or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/01/2021</td>
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<td>OCEPR</td>
<td>REV. 2</td>
<td>R2160, R216D, R21610 R2110, R2100A, R2112, R2123, R2124, R2125, R2126 R2120 R2120D R2120 DHC-6-300</td>
<td>DET 120301 Rev. 0 - Oil pressure transducer line R2000-HR2000 dated 7March 2012</td>
<td>None</td>
<td>None</td>
<td>Limitation to serial numbers 001 up to 378 included.</td>
<td>26/04/2012</td>
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<td>EASA STC no.</td>
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<td>10039182</td>
<td>LIU/THANKS TECHNIK AG</td>
<td>REV. 1</td>
<td>B737-700</td>
<td>Cabin Completion &quot;Tiger&quot;</td>
<td>The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to the certificate/ Special Conditions/CRD-02 Isolated compartments/CRI D-06 Compartment Interiors Heat Release and Smoke Emission/CRI D-10 Glass screens of displays and equipment.</td>
<td>- Compliance Record List CRDL02-1027M Issue 1- Aircraft Flight Manual Supplement A-00-10/28/1 AIFM Issue 1- Instructions for Continued Airworthiness A-00-10/28/1ICA Issue 1 for later revisions of the above listed documents approved by EASA.</td>
<td>Refer to AFM limitations on the use of Portable Electronic devices</td>
<td>24/04/2012</td>
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<td>10039188</td>
<td>DOA21 LIMITED</td>
<td>REV. 2</td>
<td>A319-115</td>
<td>Introduction of Custom Interior:Rev. 1: Correction of a typographical error concerning the Instruction(s) for Continued Airworthiness reference under Limitations/Conditions.Rev. 2: Transfer of STC holder role to DOA21 Ltd.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements. The following EASA Certification Review Items (CRIs) are part of the certification basis for this design change:-Deviation CRI D-01 Issue 1 &quot;Installation of in...</td>
<td>STC Twenty One Limited documents-Master Document List S21-MD-0866 Issue 1, dated 13th April 2012.- Aeroplane Flight Manual (AFM) Supplement S21-FMS-0060 Issue 1, dated 13th April 2012.- For documents and supplement manuals related with instructions for continued airworthiness and references on Master...</td>
<td>This design change approval is applicable on Aircraft model/AircraftSeries A319-115, aircraft serial number 402. Any subsequent change to the model of operation of the WLAN approved under this design change will require a re-examination to determine if the compliance demonstration and/ or the certification basis.</td>
<td>24/10/2023</td>
<td>Active</td>
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<td>10039195</td>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SR02344LA</td>
<td>BELL 407</td>
<td>Installation of Helias stability augmentation system and autopilot/Validation of FAA STC SR02344LA</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/ approval. FAR 27, with following amendments and related paragraphs Amtd 11: 1322; Amtd 13: 1321, 1351, 1357; Amtd 14: 29, 1581; Amtd 18: 1529; Amtd 18,...</td>
<td>-Master Data List: Helias- MCD-407 revision D dated March 19, 2012-Rotorcraft Flight Manual Supplement-Helias-RFMS-407 revision NCDated March 1, 2011-Instructions for continued Airworthiness (ICA), Document No. AA-01/001Revision B datedNovember 23, 2011for later revisions of the above listed docu...</td>
<td>VFR only</td>
<td>18/04/2012</td>
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<td>10039197</td>
<td>STRAIGHT FLIGHT, LLC</td>
<td>CESSNA 510</td>
<td>FAA STC</td>
<td>Installation of Oxygen System</td>
<td>The Certification Basis for the original product remains applicable to the certificate/ approval. The requirements for environmental protection and the associated certifications on noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Drawing List: Doc. No. FMS-4979-R007, Rev NCDated June 16, 2010-Instructions for continued airworthiness Doc No. 4979-R-007, dated June 16, 2010-Instructions for continued airworthiness Doc No. 4979-R-006, dated June 7, 2010-Revision A or later revisions of the above listed documents...</td>
<td>According limitations section of AFMS. This approval is limited to installation and use of a 40 Cubic Foot occupant Oxygen supply tank.</td>
<td>17/04/2012</td>
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<td>10039198</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>FAA STC ST01065W-D</td>
<td>BD-700-1A10/11</td>
<td>Installation of a Supplemental Crew Oxygen System</td>
<td>The Certification Basis for the original product remains applicable to the certificate/ approval. The requirements for environmental protection and the associated certifications on noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA STC No. ST01065W-D dated 30 November 2011-Midcoast Master Data List 35-261M001 Revision K, dated 30 September 2010-Airplane Flight Manual Supplement(s) 35-261M001, Revision K dated 28 March 2002 (Config. 1); or 35-261M004, Revision NC, dated 24 March 2006 (Config. 2); or 35-261M007, Revision...</td>
<td>None</td>
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<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>FAA STC No. ST02658CH-D</td>
<td>BD-700-1A10</td>
<td>Installation of a Passenger Address System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. ST02658CH-D dated 30 November 2011.</td>
<td>None</td>
<td>17/04/2012</td>
<td>Active</td>
<td>10039193</td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES NC</td>
<td>REV. 1</td>
<td>B747-400</td>
<td>B747-400 Cabin re-configuration in accordance with Northwest Aerospace Technologies Inc. Master Drawing List N0375-D10 Rev. A dated April 12, 2012</td>
<td>This certificate/approval involves a change to the requirements for environmental protection and/or a change to the certificate/noise and/or emissions levels.</td>
<td>None</td>
<td>07/05/2012</td>
<td>Active</td>
<td>10039202</td>
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<tr>
<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>A9139</td>
<td></td>
<td>EASA Validation of FAA STC No. SR02379LA &quot;Installation of a Passenger Address System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>None</td>
<td>18/04/2012</td>
<td>Active</td>
<td>10039222</td>
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<tr>
<td>ADAC LUFTFAHRT TECHNIK GmbH</td>
<td>BK117 C-2</td>
<td></td>
<td>NVIS Certification of mission equipment on an Eurocopter BK117 C2</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Flight Manual Supplement - EC145-01932-025-MFS-00 Rev. 4 or later revisions of the above listed documents approved by EASA.</td>
<td>Required pre-modification</td>
<td>EASA approval 10038241 (ECD majorchange 2237)</td>
<td>19/04/2012</td>
<td>Active</td>
<td>10039230</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>FAA STC NO. SA00530 AC-D</td>
<td>BEECH 95/B95/B95A/95A/65</td>
<td>Installation of S-TEC system 2030 single and two axis Automatic Flight Guidance Systems, model ST-802-25,30, according to bulletin N. 908 rev. 1 dated 01-Dec-2005 and with Master Drawing List N. 921083, Rev. A, dated17-Dec-2009 or later FAA approved revision (26Bv. System).</td>
<td>The Certification Basis of the original product remains applicable to this certificate/approval.</td>
<td>FAA approved S-TEC Bulletin N. 908 rev. 1 dated 01-Dec-2008, or later FAA approved revision FA/IAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N 891685, da...</td>
<td>Installer must check availability of EASA approval for one or more referenced STC when airplane is modified per that STC(s).</td>
<td>19/04/2012</td>
<td>Active</td>
<td>10039237</td>
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<tr>
<td>KUERZI AVIONICS AG</td>
<td>MF900</td>
<td></td>
<td>SATCOM Upgrade to WiFi/Activation of the WLAN option of the existing AVIATOR 300 SATCOM system MF900 S/N 118</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Document List doc No. MSL-201110583 Rev N/C dated 16April 2012 Aeronautical Flight Manual Supplement doc No. FMS-201110583 Rev 0 dated 29September 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>Limited to MF 900 S/N 118</td>
<td>19/04/2012</td>
<td>Active</td>
<td>10039239</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certificate basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AVIATION SUPPORT GmbH</td>
<td>Installation of an external Ski Basket Equipment P/N AV 115.00</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval.</td>
<td>None</td>
<td>19/04/2012</td>
<td>Active</td>
<td>10039242</td>
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<tr>
<td>ROSEN SURVIVOR SYSTEMS L.L.C.</td>
<td>Cockpit Sunvisor Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval.</td>
<td>None</td>
<td>20/04/2012</td>
<td>Active</td>
<td>10039248</td>
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<tr>
<td>HARTZELL PROPPELLER INC.</td>
<td>Installation of either a Hartzell model PHC-C3Y1-1RF/FL648A(P)-4-RPropeller and associated spinner assembly C-3535-1(P) in accordance with FAA STC SA00470AT</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval.</td>
<td>None</td>
<td>20/04/2012</td>
<td>Active</td>
<td>10039256</td>
<td></td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>Installation of S-TEC system 55/55x two axis Automatic Flight Guidance Systems, model ST-699, according to bulletin N. 689 rev.4 dated11-Jan-2010 and with Master Drawing List N. 92790. Rev. D, dated11-Jan-2010 and later FAA approved revision(14V - System).</td>
<td>The Certification Basis of the original product remains applicable to this certificate approval.</td>
<td>FAA approved S-TEC Bulletin N. 689 rev.4 dated11-Jan-2010 or later FAA approved revision(14V - System).</td>
<td>25/04/2012</td>
<td>Active</td>
<td>10039284</td>
<td></td>
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<tr>
<td>JET AVIATION AG</td>
<td>Airbus A340-600 &quot;Eminence&quot; - V/P Cabin Interior CompletionRevision 3: Transfer STC 10039249 Rev. 3 from Jet Aviation AG to Jet Aviation AG, Basel Completions.Revision 4: Transfer STC 10039249 Rev. 3 from Jet Aviation AG, Basel Completions to JET Aviation AG.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s):D-1302-002: Isolated compartments.D-1304-002: Flight crew rest compartments.D-1310-002: Displays / Monitors glass screens.D-1311-002: Cabin glass panels...</td>
<td>FAA approved S-TEC Master Drawing List No. 92790. Rev. D, dated11-Jan-2010, or later FAA approved revision(14V - System).</td>
<td>None</td>
<td>2012/2017</td>
<td>Active</td>
<td>10039289</td>
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**EASA STC no.**

- EASA STC no. 10039256 for Airbus A340-600 "Eminence" - V/P Cabin Interior Completion Revision 3: Transfer STC 10039249 Rev. 3 from Jet Aviation AG to Jet Aviation AG, Basel Completions. Revision 4: Transfer STC 10039249 Rev. 3 from Jet Aviation AG, Basel Completions to JET Aviation AG. |
<table>
<thead>
<tr>
<th>STC Holder</th>
<th>Rev. Level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tbody>
<tr>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 1</td>
<td></td>
<td>MYSTERE FALCON 20 E5, 20 GF, MF500, MF900 C550 CITATION II</td>
<td>Installation of Patch Iridium Antenna applicable to Dassault Aviation Mystere Falcon 20GF aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Use Air Enterprise Master Data List MDL-011-11 Revision 8, dated 30th March 2012 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/01/2015</td>
<td>Active</td>
<td>10039277</td>
</tr>
<tr>
<td>VAN HORN AVIATION, L.L.C.</td>
<td></td>
<td></td>
<td>206B, L, L1, L3, L4</td>
<td>Installation of Replacement Tail Rotor Blades</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. The following paragraphs of FAR 27 amendment 24-4 have been completed: 27.61; 27.65; 27.71; 27.73; 27.75; 27.76; 27.141; 27.143; 27.17...</td>
<td>-206L1-FMS-901 original issue dated 1 December 2011 - Flight Manual Supplement: 206L3-FMS-903 revision 1 dated 24 January 2012 - Flight Manual Supplement: 206L4-FMS-904 revision 1 dated 24 January 2012 - Flight Manual Supplement: or later revisions of the above listed documents approved by EASA-206-MLD.</td>
<td>When antenna holder PP-10-A441 is installed, carriage of passengers in front seats prohibited; When racks PP-10-A441 or PP-A401 are installed, a maximum of 2 passengers in near seats is permitted.</td>
<td>23/04/2012</td>
<td>Active</td>
<td>10039278</td>
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<tr>
<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td></td>
<td></td>
<td>DA40D</td>
<td>Installation of two Garmin GNS 430 W systems, two GA 35 GPS antennas, a GMA 340 audio panel, a GTX 330 Mode S transponder, a King KN-62A DME, and Cobham antennas CI 292-2 (VHF), CI 105 (SME), CI 102 (MKR)</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Flight Manual Supplement No SEAM-FM-3723/11 revision 0 dated 28/11/2011 SEAM Modification File No 3723/11 revision 1 dated 30/12/2011 or later revisions of the above listed documents approved by EASA.</td>
<td>Installation of this design change is limited to airplanes S/N D4-002 and must be performed by authorized SEAM technical personnel. This design change is approved for B-RNAV operation without coupling of the GNS 430 W systems to an autopilot. The airplane must be equipped with at least one radio naro...</td>
<td>23/04/2012</td>
<td>Active</td>
<td>10039279</td>
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<tr>
<td>FIELD AVIATION EAST, LTD.</td>
<td>REV. 2</td>
<td></td>
<td>TCCA STA C-LS41-141/D</td>
<td>Air Iceland AXivia Upgrade TCCA STA C-LS41-141/D, Issue 3</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Configuration A. DHC-8-202, SN 445 and SN 446 only; Installation must be conducted in accordance with FAA Master Drawings designated 25380000 Rev. C Flight Operations must be conducted in accordance with FAA Flight Manual Supplements as follows: 25383000, Rev IR – Collins TCAS II (Change 7) 25383001...</td>
<td>The modification is limited to aircraft serial numbers 445,446 and 450Universal UNS-1E(w) Flight Management Systems are approved for IFR route, Terminal and Non Precision approach operation. Prior to installation of this design change it must be determined that the interrelationship between this des...</td>
<td>02/03/2018</td>
<td>Active</td>
<td>10039284</td>
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<tr>
<td>SEGUNUS Inc.</td>
<td></td>
<td></td>
<td>FFA STA ST02954CH</td>
<td>Installation of a replacement pitot static probe P/N DU130-24 in accordance with FAA STA ST02954CH.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Installation Drawing EH330-24, Rev –, dated August 5, 2010 Component Installation Manual SEG130, Rev –, dated October 1, 2010; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CIF (or subsequent revisions of this decision) Instructions of Con...</td>
<td>None</td>
<td>23/04/2012</td>
<td>Active</td>
<td>10039286</td>
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<tr>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>REV. 3</td>
<td></td>
<td>747-400</td>
<td>Installation of Cobham SATCOM antenna type HQA-7001 in accordance with Master Data List MLD AE1111065L Rev 2, dated 23 April 2012.</td>
<td>The certification basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List MLD AE1111065L, rev. 2 dated 23 April 2012 AE1111065B Instructions for Continued Airworthiness, dated 14 December 2011 IPC Supplement 01388-07-06A, IPC-04 Rev BAMM Supplement 01388-07-A, AMM-01 Rev B or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/07/2015</td>
<td>Active</td>
<td>10039303</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>LUITTHAKA TECHN KAG</td>
<td>Active</td>
<td>Navigation - AG5-B Out (Automatic Dependent Surveillance Broadcast).</td>
<td>The Certification basis for the original product and the following: Additional or alternative airworthiness requirements are applicable to this certificate approval. CS25 Amrd 10. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product and remain applicable to this certificate approval.</td>
<td>Supplemental Type Definition Document, ref. STC020-1102M Issue 1, dated 27 February 2012. Aircraft Flight Manual Supplement, ref. A-00-115B-FAA Issue 01, dated 27 September 2011. Instruction for Continued Airworthiness, ref. A-00-115B-ICA Issue 01, dated 09 September 2011 or later revisions of the above.</td>
<td>None.</td>
<td>24/04/2012</td>
<td>Active</td>
<td>1030004</td>
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<tr>
<td>AMTECH AERONAUTICAL LIMITED</td>
<td>Active</td>
<td>Artex C460-N ELT and Flight Displays FDE600 Video Camera Installation.</td>
<td>The Certification basis for the original product remains applicable. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product and remain applicable to this certificate approval.</td>
<td>Amtech Aeronautical Limited (AAL) Master Data List doc. No. MDL35261 Rev. IR dated 23 May 2006. Instructions for Continued Airworthiness, doc. No. 060509203, Rev. Dated 09 May 2006. Aircraft Flight Manual Supplement doc. No. FM635281, Rev. IR dated 09 January 2012 as approved by TCCA on 14 Feb. 2012 or.</td>
<td>Limited to S/N 141</td>
<td>27/04/2012</td>
<td>Active</td>
<td>1030013</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>Installation of S-TEC system 45/50 single and two axis Automatic Flight Guidance Systems. Installation Model ST-505-48580 (28V - System) according to bulletin N. 606 rev 6 dated 21-Jan-2010 and master drawing n. 92593 rev. F, dated 21-Jan-2010 or later FAA approved revision of above documents</td>
<td>The Certification Basis of the original product remains applicable to this certificate approval. The certified noise and/or emissions levels of the original product and remain applicable to this certificate/approval.</td>
<td>FAA approved S-TEC Bulletin No.550 Rev. 6 dated 21-Jan-2010, or later FAA approved revision FAA approved revision FAA / DARPA approved Pilot Operating Handbook and/or Airplane Flight Manual Supplement, PIN 991011, dated 21-Feb-2012.</td>
<td>FAA approval.</td>
<td>25/04/2012</td>
<td>Active</td>
<td>1030017</td>
</tr>
<tr>
<td>RAISBECK ENGINEERING Inc.</td>
<td>REV. 1</td>
<td>Installation of Butterfield Industries Inspection Panel KIP/N B9501-001.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product and remain applicable to this certificate/approval.</td>
<td>FAA approved revision FAA approved revision FAA / DARPA approved Pilot Operating Handbook and/or Airplane Flight Manual Supplement, PIN 991011, dated 21-Feb-2012.</td>
<td>FAA approval.</td>
<td>06/08/2018</td>
<td>Active</td>
<td>1030022</td>
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<tr>
<td>AIRCRAFT TECHNICAL DEVELOPMENT Inc.d/b/a AEROCON ENGINEERING COMPANY</td>
<td>REV. 2</td>
<td>Installation of On-Board AirStart. This modification installs an On-Board AirStart. This STC installs into an aircraft with a new configuration (new aircraft model).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product and remain applicable to this certificate/approval.</td>
<td>Master Drawing List, Ref. 1623-7000 Revision F Instruction for Continued Airworthiness, Service and Operational Manual. Ref. MMS-A340-1462 revision C Lift Module Instruction for Continued Airworthiness, Service and Operational Manual. Ref. MMS-A340/FT-1713 AirStart Weight and Balance Supplement, Rel...</td>
<td>Intergration of this STC with cabin arrangement features has not been evaluated. Should this STC be part of a cabin arrangement, the interactions between this STC and any cabin or seating arrangement must be evaluated for compliance with applicable certification requirements in the frame of the STC(s).</td>
<td>06/02/2013</td>
<td>Active</td>
<td>1030027</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>10039335</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td></td>
<td>F2000EX EASY</td>
<td>TCAS 2000 upgrade to TCAS change 7.1 on F2000EX EASy</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>DF 5 Change Approval Sheet DSF5049-CAS Issue A dated 07 March 2012</td>
<td>As per DF 5 Change Approval Sheet DSF5049-CAS Issue A dated 07 March 2012</td>
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<tr>
<td>10039339</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>REV. 3</td>
<td>TCCA STC SA11-27, ISSUE 5</td>
<td>DHC-8-301, -311, -314, -315, DHC-8-201, -202, DHC-8-102, -103, -106</td>
<td>Universal EFI-860R EFIS Cockpit Upgrade Inclusion of MMS 2503110 Rev. F5, corrections of minor typographical errors, and editorial changes.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>DHC-8-100 and 200 Series Installation must be performed in accordance with FACI Master Drawing/ Data List designated 2500000 Rev K. Flight Operations must be performed in accordance with FACI Flight Manual Supplements listed as follows: 2500000 Rev B, EFI-860R Electronic Flight Instrument Suite. Prior to installation...</td>
<td>25/10/2013</td>
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<tr>
<td>10039340</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>REV. 1</td>
<td>TCCA STC C-LSA12-014D</td>
<td>DHC-8-202</td>
<td>Installation of upgraded Cabin Floor Panels/TCCA STC C-LSA12-014D, Issue 1</td>
<td>FAR 25 as per EASA M.A.191, Issue 5 (incl. EWS). The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation must be in accordance with TCCA Certification basis FieldAviation/Company Inc. Master Data List 2500000, Rev. A, dated 23 April 2012. Maintenance is unchanged from the instructions provided in the Bombardier DHC-8-200 Maintenance Manual PSM 1.0-2.1. Zonal inspections are unchanged from the manual.</td>
<td>The modification is limited to Serial numbers 445, 446 without 1996/992/Cockpit inserts. Only: Floor loads of 75 kN/ft² are unchanged from the original product. Underfloor support structure modified per Bombardier CR or CR3200000000 or equivalent.</td>
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<tr>
<td>10039353</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>A321-212/-213/-231/-232 A320-211/-212/-221/-222/-223 A319-111/-112-131/-132/-133 A319-111/-112-113-114-115</td>
<td>Replacement of Lilton MK-1 Global Positioning System Sensor Unit (GPS/INS) by CMC Electronics GPS/INS CMA-5024 (STC revised for validity extension to A319 and A321 models)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>None</td>
<td>None</td>
<td>None</td>
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<tr>
<td>10039356</td>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td>A340-310, A340-311, A340-312, A340-313 A340-210, A340-211, A340-212, A340-213</td>
<td>Camera Unit and structural provision installation This STC installs one camera on the tail of the aircraft, watching forward. This covers the mechanical installation of the camera unit and the penetration of the pressure.</td>
<td>The certification basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List, BA34031701-001 Issue A, dated 26.04.2012 for Continuous Airworthiness. BA34031701-150-01 Issue A dated 26.04.2012 for later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>None</td>
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<tr>
<td>10039357</td>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td>A340-310, A340-311, A340-312, A340-313 A340-210, A340-211, A340-212, A340-213</td>
<td>Tailwind SS6560 antenna installation This STC installs a tailwind SAT TV antenna and radome on top of the fuselage.</td>
<td>The certification basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List, BA34035203-001 Issue A, dated 26.04.2012 for Continuous Airworthiness. BA34035203-150-01 Issue A dated 26.04.2012 for later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>None</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td>Attachment provisions on Falcon 50 S/N 011.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certificated noise/ or emissions levels of the original product unchange and remain applicable to this certificate/ approval.</td>
<td>Master Document List N: PP-033-MDL-01 Issue 1 dated 28.04.2012 Instructions for Continued Airworthiness N: PP-033-ICA-00 Issue 0 dated 01.04.2012 or later revisions of the above listed documents approved by EASA.</td>
<td>Limited to S/N 011.</td>
<td>30/04/2012</td>
<td>Active</td>
<td>10059373</td>
<td></td>
</tr>
<tr>
<td>ISOLAIR Inc., ISOLAIR HELICOPTER BY</td>
<td>Title: Installation of 3900-480 Spray System Description: The installation of the 3900-480 Spray System pertains to the installation of an agricultural spray system onto the Enstrom480 and 480B rotorcraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certificated noise/ or emissions levels of the original product unchange and remain applicable to this certificate/ approval.</td>
<td>Master Document List N: 3900-480PSM- EASA; Chapter 4 &quot;Airworthiness Limitations&quot; of the Instructions For Continuation Airworthiness, ICA Manual No. 3900-480- EASA ICA, Rev. A dated 7th March 2012 or later revisions of the above listed documents approved by EASA.</td>
<td>Restricted to agricultural use.</td>
<td>02/05/2012</td>
<td>Active</td>
<td>10059386</td>
<td></td>
</tr>
<tr>
<td>DUNLOP AIRCRAFT TYRES LIMITED</td>
<td>Installation of Dunlop Aircraft Tyres on ATR 42-200-300-320-400-500.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certificated noise/ or emissions levels of the original product unchange and remain applicable to this certificate/ approval.</td>
<td>Master Document List, ref: KT509009MDL, dated 11 April 2012 Instructions for Continued Airworthiness, ref: KT509009ICA, dated 11April 2012 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>02/05/2012</td>
<td>Active</td>
<td>10059387</td>
<td></td>
</tr>
<tr>
<td>ASI INNOVATION</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certificated noise/ or emissions levels of the original product unchange and remain applicable to this certificate/ approval.</td>
<td>None</td>
<td>The EFD 1000 must not be used as primary flight and navigation source. Independent and dissimilar instruments must be installed for primary/primarilization of altitude, attitude, airspeed, vertical speed, magnetic heading and navigation data, as required for day and night VFR flights.</td>
<td>02/05/2012</td>
<td>Active</td>
<td>10059405</td>
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<td>SPECTRUM AEROMED, LLC/ SIA SPECTRUM AEROMED</td>
<td>To Install an Air Ambulance Conversion. This Change instructs the interior and electrical changes to the aircraft. Stretchers and medical equipment installation will be certified by the appropriate vendor.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certificated noise/ or emissions levels of the original product unchange and remain applicable to this certificate/ approval.</td>
<td>Installed in accordance with Spectrum Aeromed Inc Master Document List 8033-3DL, Revision 3, dated September 26, 2011. Operation in accordance with Spectrum Aeromed Inc Approved Flight Manual Supplement 6033-2FL, Revision 2, dated September 15, 2011. Maintained in accordance with Instructions for Continuation Airworthiness Limitations.</td>
<td>17/04/2019</td>
<td>Active</td>
<td>10059407</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10039421</td>
<td>BRITISH AIRWAYS PLC</td>
<td>REV. 2</td>
<td>A321-200, A320-214, A330-200, A350-900</td>
<td>AIRFILM TCAV Change 7.1 Updated Rev. 1</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25 Amendment 11 for the requirements affected by this change.</td>
<td>- Master Data List AFM-FAA/AM-029-10, Rev. 1 dated August 1, 2011- Installation Instructions AFM-FAA/AM-006, dated October 7, 2011</td>
<td>None</td>
<td>19/10/2017</td>
</tr>
<tr>
<td>10039428</td>
<td>AIRFILM CAMERA SYSTEMS, INC</td>
<td>REV. 1</td>
<td>FAA STC SR02376LA</td>
<td>Airfilm Nose Utility Mount</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>- Master Drawing List AFM-FAA/AM-029-10, Rev. 1 dated August 1, 2011- Installation Instructions AFM-FAA/AM-006, dated October 7, 2011</td>
<td>None</td>
<td>03/06/2012</td>
</tr>
<tr>
<td>10039429</td>
<td>AIRFILM CAMERA SYSTEMS, INC</td>
<td>REV. 2</td>
<td>FAA STC SR02377LA</td>
<td>Airfilm Tail Utility Mount</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>- Master Drawing List AFM-FAA/AM-029-10, Rev. 1 dated August 1, 2011- Installation Instructions AFM-FAA/AM-006, dated October 7, 2011</td>
<td>None</td>
<td>03/06/2012</td>
</tr>
<tr>
<td>10039436</td>
<td>JET AVIATION AG, BASEL</td>
<td>REV. 2</td>
<td>A340-300</td>
<td>Airbus A340-300 &quot;0Star&quot; VIP Cabin Interior Completion/This STC installs a VIP cabin interior. The Revision 1 removed a limitation. The Revision 2 has introduced two configurations regarding the Crew Rest Compartment utilisation.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): D-2302-001, Isolated Compartment(s): D-2304-001, Crew Rest Compartment(s): D-2310-001, Glass Screens of Displays/ Monitors(s): D-2311-001, Installation of Glass Pan...</td>
<td>Master Data List, Ref: B3400955-001 Revised H, dated 17 July 2013; Airplane Flight Manual Supplement, Ref: B3400955-201-01 Revision E, dated 28 May 2013; Instructions for Continued Airworthiness B3400955-100-01 Revision A, dated 26 April 2013; Night and Balance Manual Supplement, Ref: AFG-15M, dated 09/10/2012</td>
<td>Aristair installation is not part of the approved configuration</td>
<td>19/07/2013</td>
</tr>
<tr>
<td>10039438</td>
<td>MECAER AVIATION GROUP S.p.A.</td>
<td>REV. 1</td>
<td>AW100SP</td>
<td>Installation of Cabinet Five Star P/N 6A100AD002-001 in AW100SP Modification in accordance with MECAER Notice of Design Change NDC6010AD-003 Issue A dated 24/04/2012</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>*Owner's Manual for Cabinet Five Star Installation P/N 6A100AD002-001 P/N 6A100RNM-9-2 First Issue Rev. A dated 04/03/2012 or later approved revision Notice of Design Change Ref. NDC 6A100AD-003 Issue A dated 24/04/2012 <em>AW100SP Cabinet Five Star Compliance Checklist</em> Ref. 6A100RAD-01 Rev. D dated 24/...</td>
<td>This STC approval does not include the installation of the in-flight entertainment equipment inside the Cabinet Five Star.</td>
<td>16/05/2012</td>
</tr>
<tr>
<td>STC STC No.</td>
<td>STC Holder</td>
<td>Type</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10039454</td>
<td>EUROCOPTER</td>
<td>CLAS UK RESPONSIBILITIES</td>
<td>Installation of Kelvin 275 Camera per Drawing List DL/1352044-001, Issue A</td>
<td>The Certification Basis for the original product remains applicable with certificate/approval. The requirements for environmental protection and the associated certificate/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Rotordraft Flight Manual Supplement EC 135 P1 Supplement No. 5A, Issue 1; later revisions of the above listed documents approved by EASA.</td>
<td>Day/night VFR operation only. Operation of the Kelvin 275 camera approved under this STC (EC UK/135/2044) has to be done, in conjunction with EC UK/135/2045 and EC UK/135/2043.</td>
<td>04/05/2012</td>
<td>Active</td>
</tr>
<tr>
<td>10039466</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SR0281NY</td>
<td>Installation of Hardpoint and Spotter Fixtures in accordance with Aeronautical Accessories Inc. Installation Drawing List for Hardpoint Fixtures Kit, Ceiling Spotter Fixture Kit, Floor Spotter Fixture Kit, Report Number AA-0943, Rev C, dated June 29, 2010.</td>
<td>The Certification Basis for the original product remains applicable with certificate/approval. The requirements for environmental protection and the associated certificate/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Aeronautical Accessories Inc. Installation Drawing List for Hardpoint Fixtures Kit, Ceiling Spotter Fixture Kit, Floor Spotter Fixture Kit, Report Number AA-0943, Rev C, dated June 29, 2010; - Lawon Aerospace LLC Instructions for Continued Airworthiness (CAW) HELICOOM V2 ANA, HELICOOM V2+ ANA, HELICOOM V2+ ANA PANEL Edit A Revision 04 - Flight Manual Supplement Revision 04 - Flight Manual Supplement Revision 07/2015.</td>
<td>None</td>
<td>07/06/2012</td>
<td>Active</td>
</tr>
<tr>
<td>10039479</td>
<td>SAFRAN ELECTRONICS &amp; DEFENSE</td>
<td>REV. 2</td>
<td>AS 355 F, AS 355 F1 AS 355 F2, AS 355 N HELICOM V3 ANA / V2+ ANA HELICOM V2+ ANA</td>
<td>The Certification Basis for the original product remains applicable with certificate/approval. The requirements for environmental protection and the associated certificate/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>NB-AS355 ANA-HV2 ANA OH PANEL Edit A Revision 04 - Flight Manual Supplement Revision 04 - Flight Manual Supplement Revision 07/2015.</td>
<td>The system does not replace standard maintenance records and providing Prior to installation of this design change must be determined that the operational between this design change and any other previously installed design change and/or repair will introduce roadworthy effect upon the a...</td>
<td>27/02/2020</td>
<td>Active</td>
</tr>
<tr>
<td>10039495</td>
<td>GFD GmbH</td>
<td>REV. 2</td>
<td>35A/36A 35A, 36A</td>
<td>Enhanced Surveillance Upgrade to MST ITA Transponder (Honeywell) Rev. 2, Transfer to GFD GmbH.</td>
<td>The Type and OSD Certification Bases (CB) for the original productremains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AEC Master Data List: ref 5187-049, rev A, dated 17 April 2012; AEC AFM Supplement: ref 5187-071, rev R, dated 30 March 2012. AEC Instructions for Continued Airworthiness: ref 5187-085, rev R, dated 23 March 2012; later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will not introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>GFD GmbH</td>
<td>36A</td>
<td>Replacement of UMS-1M or UMS-1B by UMS-1W MMS/Rev. 2. Transfer to GFD GmbH</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>According to AEC Master Data List No. 5183-049, Rev. A, or 5183-049_01, Rev. IR. To be installed i.a.w. AEC drawing list 5183-082, Rev. B or 5183-082-01, Rev IR. To be operated i.a.w. AEC AFMS 5183-071, Rev. B or 5183-071-01, Rev IR. If 5183-071-02, Rev IR. To be maintained i.a.w. AEC ICA Document No. 5183.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse affect upon the airworthiness of the product.</td>
<td>23/08/2012</td>
<td>Active</td>
<td>10039492</td>
</tr>
<tr>
<td>GAMA ENGINEERING LTD</td>
<td>BN.2A MARK III TRLISLANDER, BN.2A MARK III TRLISLANDER, BN.2A MARK III TRLISLANDER, BN.2A MARK III-1 TRLISLANDER, BN.2A MARK III-2 TRLISLANDER,</td>
<td>Installation of Aspen EFD1000 Electronic Primary Flight Display</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/ approval CS23 Amendment 2. The requirements for environmental protection and the associated certificate/ approval remain applicable to this certificate/ approval.</td>
<td>Master Data List MDL-1282 Issue 1, dated 04 May 2012. Change Statement CS-1282 Issue 1, dated 04 May 2012. Flight Manual Supplement AFMS-1282 Issue 1 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>08/05/2012</td>
<td>Active</td>
<td>10039502</td>
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<tr>
<td>MACH TECHNICAL DESIGN B.L.</td>
<td>BELL 412</td>
<td>Surveillance Camera Installation according to MDL-FAA-0110 Rev. B. (The installation consists in the installation of a camera to capture images and the associated equipment to process and send them to a groundstation.)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate/ approval remain applicable to this certificate/ approval.</td>
<td>Rotorcraft Flight Manual Supplement FAA-0110, Issue 1, dated 02/03/2012 or later revisions of the above listed documents approved by EASA-MDL-FAA-0110, Rev B dated 07/05/2012-OT- FAA-0110, Rev 00, dated 19/12/2011-DIP-FAA-0110, Rev 0, dated 01/12/2010</td>
<td>The installation of this STC is compatible with the installation of the TCCA STC BHA-57 (STA RC 1182); Installation of Heli-Access Steps PHIN D205-0633-011 in accordance with Dart Aerospace Ltd. Drawing IN-D205-633, Rev A. Prior to installation of this modification it must be determined that there is no adverse affect upon the airworthiness of the product.</td>
<td>08/06/2012</td>
<td>Active</td>
<td>10039505</td>
</tr>
<tr>
<td>TELEDYNE UK LIMITED</td>
<td>767-200, 767-300, 767-300F</td>
<td>Installation of Wireless Quick Access Recorder/Ref. 2. Transfer from Teledyne Limited to Teledyne UK Limited.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate/ approval remain applicable to this certificate/ approval.</td>
<td>Teledyne Controls Master Data List STC-MDL-013-15 Issue 15 dated 04 January 2012 (&lt;=) or later revisions of the above listed document(s) approved acceptably under the EASA system.</td>
<td>The installation of this change/repair must be determined that there is no adverse affect upon the airworthiness of the product.</td>
<td>26/11/2012</td>
<td>Active</td>
<td>10039512</td>
</tr>
<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>P2006T</td>
<td>Police cabin configuration/Installation of a camera IR for 12” grilles, operation console and replacement of hatch doors by fixed cover(s) or standard analogue instruments equipped aircraft step 2 is applicable</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate/ approval remain applicable to this certificate/ approval.</td>
<td>Depending on aircraft instrument panel configuration (analogue instruments or Garmin G950) the following documents are required: “Police cabin configuration”, Doc. No. STC-AT-045, Rev. 2; “Installation Instructions”, Doc. No. 945-B-STC-AT, Rev. 04F; Supplements Doc. No. AFM-1-STC-AT-045, Rev. 0 or AFM2.</td>
<td>Prior to installation of this change/repair must be determined that there is no adverse affect upon the airworthiness of the product.</td>
<td>08/06/2012</td>
<td>Active</td>
<td>10039513</td>
</tr>
<tr>
<td>SAFRAN CABIN BELLINGHAM, INC.</td>
<td>FAA ST02237SE</td>
<td>Installation of 2 new avionics racks in the EJ bay of a Boeing 747-400 series aircraft in accordance with Heath Tecna Inc. Master Data List PD-162 Rev. A, dated April 12, 2012 or later approved revision.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/ approval:</td>
<td>Heath Tecna Inc. Master Data List PD-162 Rev. A, dated April 12, 2012; Instructions for Continued Airworthiness Heath Tecna Inc. Doc HTM 1004 Revision B, dated April 4th, 2012 or later revisions of the above listed document(s) approved by EASA.</td>
<td>Approval of this change applies to Boeing 747-400 aircraft listed in FAA-8080-162 Rev. A, dated April 12, 2012 or later approved revisions only.</td>
<td>10/06/2012</td>
<td>Active</td>
<td>10039535</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10039546</td>
<td>ROEDER PRAEZISION GMBH</td>
<td>TULB06F</td>
<td></td>
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<td>Installation of an additional Alternator with Battery for Mission Purposes</td>
<td>Flight Manual Supplement 01271AM4200030-01Aur later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>10/05/2012</td>
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<tr>
<td>10039554</td>
<td>ROEDER PRAEZISION GMBH</td>
<td></td>
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<td>Installation of provision for the installation of a camera in the lower fuselage as described in description document 14576DE E300 10 01D.</td>
<td>Flight Manual Supplement 01271CA4200030-01Eng in Effecting Order 01271E0400300-01A</td>
<td>None</td>
<td>10/05/2012</td>
</tr>
<tr>
<td>10039562</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>Installation of a galley in accordance with SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA) Master Document List+A-11-03-TwinJet, GalleyII, MD/04, Edition 1 April 2012</td>
<td>Flight Manual Supplement 01271CA4200030-01Eng in Effecting Order 01271E0400300-01A</td>
<td>None</td>
<td>05/06/2012</td>
</tr>
<tr>
<td>10039675</td>
<td>AIRCRAFT CABIN MAINTENANCE GMBH</td>
<td>REV. 1</td>
<td></td>
<td>3-B767 2-B737 1-A330</td>
<td>Installation of refurbished Passenger Seat Bell P/N EPA20511-2-01-20 on 16g seats</td>
<td>Flight Manual Supplement 01271CA4200030-01Eng in Effecting Order 01271E0400300-01A</td>
<td>None</td>
<td>21/05/2012</td>
</tr>
<tr>
<td>10039579</td>
<td>NEW SYSTEMS ENGINEERING</td>
<td>REV. 1</td>
<td>FAA STC ST10977SC</td>
<td>DHC-8-301-311/314/315 DHC-8-201-202 DHC-8-102-103-106</td>
<td>Installation of Dual Universal Avionics Flight Management Systems (FMS) with Global Positioning System (GPS)/FAA STC ST10977SC</td>
<td>Flight Manual Supplement 01271CA4200030-01Eng in Effecting Order 01271E0400300-01A</td>
<td>None</td>
<td>21/05/2012</td>
</tr>
<tr>
<td>10039607</td>
<td>AVIDYNE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SA0282B0</td>
<td>PA-46R-350T</td>
<td>Integrated Flight Display</td>
<td>Flight Manual Supplement 01271CA4200030-01Eng in Effecting Order 01271E0400300-01A</td>
<td>None</td>
<td>11/10/2012</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>365 AEROSPACE DIDS LIMITED</td>
<td>REV.</td>
<td>25 SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Donnic Digital Media Reproducer</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification of DN-DM72-23-000 Issue 17 dated 19th April 2019 Instructions for Continued Airworthiness V.07 dated 12 January 2019ON-MD90-23-0000-01 V.02 Installation of DMR onto MD-80-30 (Eva Air) DN-737-23-0000-02 V.02 Installation of DMR onto Boeing 737 (Neo) ON-767-23-0000-03 V.01 Installation of DMRD...</td>
<td>Amendment to the STC is required in order to extend the applicability of the STC to the following types, A318 (EASA.A.084), B747(EASA.A.196). The amendment to the STC shall provide the installation instructions and ICA to supplement the common data package that is defined within Modification DN-DM72-23-000...</td>
<td>02/12/2019</td>
<td>Active</td>
</tr>
<tr>
<td>AEROSPACE DESIGN AND CERTIFICATION LTD d. T/A AERODAC</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B200 KING AIR</td>
<td>Installation of polyurethane reticulated foam into fuel tanks, as defined by JN233 ADC17 (1) Master Data List</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificate/approval of the noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with JN233 ADC29 (1) Issue 1(1)&lt;(&gt;)Modification Instructions for Continued Airworthiness Operations in accordance with JN233 ADC54-23 (1) Issue 1(1)&lt;(&gt;)&gt;Instructions for Continued Airworthiness (CAA) or later revisions of the abo...</td>
<td>This STC grants no credit for any fuel tank flammability reduction.</td>
<td>14/05/2012</td>
<td>Active</td>
</tr>
<tr>
<td>ACS-NAI Ltd.</td>
<td>TCAS STC C-LSAS05-3165D</td>
<td>CL-600-2B16</td>
<td>Aircraft Completion ASN 5618, 12 Passenger Interior Configuration.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCAC STC No. C-LSAS05-3165D Issue 2, dated 21 July 2006 Aeronautical Services Ltd Modification Data Summary 610000 Revision A, dated 18 July 2006 Midcoast Aviation Airplane Flight Manual Supplement 34-424M005 Revision A, dated 17 July 2006 Midcoast Aviation Airplane Flight Manual Supplement 34-261M017 Revision A, dated 21 July 2006 Aero Consulting Services Ltd Supplement 34-424M005 Revision A, dated 15 July 2006</td>
<td>Restricted to ASN 5618</td>
<td>15/06/2012</td>
<td>Active</td>
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<tr>
<td>AIRBUS HELICOPTERS DEUTSCHLAND GmbH</td>
<td>LBA STC EMZ NR. 0643/0588</td>
<td>BO 105 LS A-3</td>
<td>Title: Superlifter Conversion Kit 105-800331 Description Original STC: The model B0195 LS A-3 is changed in accordance withdrawing no. 105-800331. The changed model is called “Superlifter”. Revision 1: The Airworthiness Limitations related to EMZ No. 0654/3058 has been changed and approved.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Original STC-Flight Manual BO 105 LS A-3, DOT, approved dated 26th February 1995 (later DOT approved versions until 29th June 2009) A.196. Appendix for BO 105-LS A-3 SuperlifterOperation*, LBA approved dated 31st October 1995 or later LB Approved versions until 27th September 2019.</td>
<td>The implementation of the modified cargo hook 105-800331 is only approved after conversion of the helicopter in accordance with drawingno. 105-800331. This STC is valid in conjunction with the EASA TCDS R.011 in the applicable version. This Supplemental Type Certificate is approved only for the products...</td>
<td>24/06/2012</td>
<td>Active</td>
</tr>
<tr>
<td>AERODATA AG</td>
<td></td>
<td>B300</td>
<td>Installation of Aerodata Automatic Flight Inspection System AD-AFIS-0270</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval CS-23 Amendment 2 (not required by 21A.101) The requirements for environmental protection and the associated certificated noise and/or emissions...</td>
<td>Flight Manual Supplement: AIMF-135, Rev for later revisions of the above listed documents approved by EASA Installation Manual,... HB-AC-1118, Rev 8 Master Document List,... MDC-AC-0177, Rev 3 Maintenance Supplement, AIMM-1117, Rev 0</td>
<td>None</td>
<td>15/06/2012</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>10039676</td>
<td>AIRBUS A.S.</td>
<td>A320-214</td>
<td></td>
<td>Airbus Corporate Jet Centre Modification C31100 - Issue 1 - Installation of Business Class Seating Area - &quot;Gulf Air - Type A&quot; A320-214</td>
<td>Modification Approval Sheet - ref.: C31100&lt;+&gt; Issue 1, dated 16 May 2012 - Aircraft Flight Manual AFM Supplement&lt;+&gt; - Issue 1, dated 09 May 2012</td>
<td>16/05/2012</td>
<td>Active</td>
<td>10339676</td>
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<tr>
<td>10039667</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 1</td>
<td>208, 208B</td>
<td>Installation of 5-blade MT-Propeller Model MTV-271 - E-C-F-R/CFR5455a on Cessna 208, 208B equipped with an electric de-icing system.</td>
<td>According to MT-Propeller Master Documentation List No. E-2110, Rev 0To be operated in accordance with MT-Propeller AFM-S Document No.E-2094, Rev 1or later revisions of the above listed documents approved by EASA. To be installed i.a.e. Installation Instruction Doc. No. E-2095, Rev 2or later Revisions.</td>
<td>21/12/2016</td>
<td>Active</td>
<td>10339667</td>
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<tr>
<td>10039669</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td>AW106SP</td>
<td></td>
<td>Installation of In-flight Entertainment System (IES) (composed by KAPIN KSxU-419 and KNAV-442)</td>
<td>Modification Bulletin N04-142 Rev. 0, dated May 2012.</td>
<td>16/06/2012</td>
<td>Active</td>
<td>10339669</td>
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<td>10039703</td>
<td>ACS-NAI Ltd.</td>
<td>TCCA STC NO. C- LSA12-038/D</td>
<td>BD-700-1A10</td>
<td>Installation of Complete Custom Aircraft Interior - ASN 9434</td>
<td>Restricted to ASN 9434. The Wireless Local Area Network (WLAN) system is intended to provide Internet connectivity and email services to the airplane's cabin using portable devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis.</td>
<td>18/06/2012</td>
<td>Active</td>
<td>10339703</td>
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<tr>
<td>10039704</td>
<td>EUROCOPTER UK Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>EC 135 P1</td>
<td>Cabin Configuration Installation of equipment rack and pedestal mounted LCD monitor in the forward cabin, as per ECD design change EC UK/135/2012 A330-300, MSN 4827, MSN 4860, MSN 4865</td>
<td>None.</td>
<td>18/05/2012</td>
<td>Active</td>
<td>10339704</td>
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<tr>
<td>10039705</td>
<td>SABENA TECHNICS BGC</td>
<td>A330-223 / -243</td>
<td>A330-201 / -202 / -203</td>
<td>Panasonic eXPhone System activation on A330 with eXConnect and Rack P/NRD-04690210This STC installs a Panasonic eXPhone System allowing, in conjunction with other devices, the use of cellphones during flight.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the above listed documents approved by EASA. To be installed i.a.e. Installation Instruction Doc. No. E-2110, Rev 2or later Revisions.</td>
<td>18/05/2012</td>
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<td>STC Holder</td>
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<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC ST02923CH</td>
<td>18/05/2012</td>
<td>00ER</td>
<td>Installation of ECS EZMountTM/Mechanical Mounting Assemblies</td>
<td>Validation of FAA STC ST02923CH, EFB</td>
<td>- Master Data List, Doc. No. ECS-200976, Revision C, dated 26 January 2011; AFM Supplement ECS-200977, Initial release, dated 11 February 2011; EWIS Assessment ECS-200978, Revision A, dated 26 January 2011; JCA ECS-200923, Revision A, dated 7 January 2011; - Weight and Balance Manual ECS-200971; Revision 2. Limited to S/N 6414</td>
<td>18/05/2012</td>
<td>Active</td>
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<tr>
<td>HELICOPTERS ITALIA S.R.L.</td>
<td>AS 365N2</td>
<td>18/05/2012</td>
<td>00ER</td>
<td>Installation of Audio Voice Warning on Radar Altimeter Thales Avit-8(Af). Change Hi-365-34-01.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Helicopters Italia Rotorcraft Flight Manual Supplement H365-34-01. FMS, Rev. 1, dated 19th January 2012a later revisions of the above listed document approved by EASA. Helicopters Italia. Master Document List (MDL) H365-34-01.MDL Rev. 2, dated 14th February 2012. Helicopters Italia. Install...</td>
<td>18/05/2012</td>
<td>Active</td>
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<tr>
<td>JET AVIATION AG, BASEL</td>
<td>FALCON 2000</td>
<td>24/05/2012</td>
<td>00ER</td>
<td>WLAN activation For WLAN working in the 2.4 Gt frequency band</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, Doc. No. BF2TH23262-001, Rev. IR, 20.04.2012; AFMS, Doc. No. BF2TH2362-001-01, Rev. IR, 29.02.2012- ICA, Doc. No. BF2TH23262-150-01, Rev. IR, 27.02.2012 later revisions of the above listed document approved by EASA.</td>
<td>24/05/2012</td>
<td>Active</td>
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<td>SPECTRUM AEROMED, LLC d/b/a SPECTRUM AEROMED</td>
<td>REV. 1 FAA STC SR02968CH</td>
<td>26/04/2019</td>
<td>00ER</td>
<td>Installation of equipment rails and rear ECS-209277, Revision2...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. RSG AeroDesign Master Document List 00A02160002(+)&lt;j&gt; No Revision, dated February 23 2011; 2. RFM Supplement 19A0210001, dated June 20 15 2013. Instructions for Continued Airworthiness, Document No. 16A02160002 No Revision, dated April 201011 or later revisions of the above listed document(s) appro...</td>
<td>26/04/2019</td>
<td>Active</td>
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<tr>
<td>SPECTRUM AEROMED, LLC d/b/a SPECTRUM AEROMED</td>
<td>REV. 1 FAA STC SR02961CH</td>
<td>26/04/2019</td>
<td>00ER</td>
<td>Installation of a medical seat, fire extinguisher, utility track and cylinder mount.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. RSG AeroDesign Master Document List 00A02390001, No Revision, dated February 23 2011; 2. RFM Supplement 19A02390001, dated June 20 15 2013. Instructions for Continued Airworthiness, Document No. 16A02390001 No Revision, dated March 4 2011 or later revisions of the above listed document(s) appro...</td>
<td>26/04/2019</td>
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<td>STC Holder</td>
<td>STC No.</td>
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<td>SPECTRUM AEROMED, LLC/ b/a SPECTRUM AEROMED</td>
<td>10039718</td>
<td>REV. 2</td>
<td>FAA STC SR02974Ch</td>
<td>A1005/ A1005P</td>
<td>Installation of an interior panel</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchangeable and remain applicable to this certificate/approval.</td>
<td>1. Spectrum Aeromed Master Drawing List 7011DL, Revision 4. dated 29 May 2014, or later approved revisions. 2. Flight Manual Supplement 7011- FMS, Revision 3h. (&gt;) 2011.4. dated 30 July 2014, or later approved revisions. 3. Instructions for Continued Airworthiness, Manual 7011PM, Revision 5. dated 29 May 2014.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect on the original product.</td>
</tr>
<tr>
<td>SPECTRUM AEROMED, LLC/ b/a SPECTRUM AEROMED</td>
<td>10039719</td>
<td>REV. 1</td>
<td>FAA STC SR02978Ch</td>
<td>A1005P</td>
<td>Installation of an interior panel</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchangeable and remain applicable to this certificate/approval.</td>
<td>1. RSG Aerodesign Master Document List 06A02240001, No Revision, dated February 23h. (&gt;) 2011.2. dated April 19, or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect on the original product.</td>
</tr>
<tr>
<td>SPECTRUM AEROMED, LLC/ b/a SPECTRUM AEROMED</td>
<td>10039720</td>
<td>REV. 1</td>
<td>FAA STC SR02970Ch</td>
<td>A1005P</td>
<td>Installation of a medical floor line: EASA STC 10039720 from SNADY CREEK ENTERPRISES, LLC b/a Spectrum Aeromed to Spectrum Aeromed, LLC b/a Spectrum Aeromed.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchangeable and remain applicable to this certificate/approval.</td>
<td>1. RSG Aerodesign Master Document List 06A02160003, No Revision, dated February 23 2011h. (&gt;) 2011.2. dated April 19, or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect on the original product.</td>
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<tr>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY, INC/ USA, INC.</td>
<td>10039721</td>
<td>REV. 1</td>
<td>FAA STC SR02989Ch</td>
<td>A1005P</td>
<td>Installation of an interior panel</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchangeable and remain applicable to this certificate/approval.</td>
<td>1. RSG Aerodesign Master Document List 06A02160031, No Revision, dated February 23 2011h. (&gt;) 2011.2. dated April 19, or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect on the original product.</td>
</tr>
<tr>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY, INC/ USA, INC.</td>
<td>10039722</td>
<td>REV. 1</td>
<td>FAA STC ST00013B</td>
<td>EMB-135BJ</td>
<td>Installation of an Enfite Refrigerator</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchangeable and remain applicable to this certificate/approval without any im...</td>
<td>-Master Document List ref A1005/ A1005P. dated 30 July 2014, or later approved revisions.</td>
<td>None</td>
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List of EASA Supplemental Type Certificates

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<td>System Record</td>
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<td>10039731</td>
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<td>F49.25 as per EASA M-A.191, Issue 5 (incl. EWIS).The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/required noise levels and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The modification is limited to Serial numbers 435, 440 and 441.</td>
<td>21/05/2012</td>
<td>Active</td>
<td>10039731</td>
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<tr>
<td>10039755</td>
<td>REV. 2</td>
<td>Installation of Tech-Tool Plastics, Inc. Master Drawing List No. BK117-100-100, dated 03/07/2018.</td>
<td>None.</td>
<td>15/06/2018</td>
<td>Active</td>
<td>10039755</td>
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<td>ROEDER PRAE2VISION GMBH</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated/required noise levels and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this modification it must be determined if there is an adverse effect upon the airworthiness of the product.</td>
<td>02/05/2012</td>
<td>Active</td>
<td>10039763</td>
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<tr>
<td>10039783</td>
<td>REV. 1</td>
<td>Installation of Tech-Tool Plastics, Inc. Master Drawing List No. BK117-100-100, dated 03/07/2018.</td>
<td>None.</td>
<td>22/05/2012</td>
<td>Active</td>
<td>10039783</td>
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<td>TECH-TOOL PLASTICS, Inc.</td>
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<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated/required noise levels and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>22/05/2012</td>
<td>Active</td>
<td>10039764</td>
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<tr>
<td>10039764</td>
<td></td>
<td>Installation of Aircraft Completion Modifications - ASIN 9440.</td>
<td>Restricted to ASIN 9440. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane’s cabin using portable electronic devices (PEDs). Any other intended function of this equipment is required to be examined on the certification basis.</td>
<td>22/05/2012</td>
<td>Active</td>
<td>10039764</td>
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<tr>
<td>BOMBARDIER INC.</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated/required noise levels and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>23/05/2012</td>
<td>Active</td>
<td>10039761</td>
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<tr>
<td>10039781</td>
<td></td>
<td>Installation of Airfoil Pacific Belly Utility Mount AP206UM02-10, Validation of FAA STC SR01688LA.</td>
<td>- This Belly utility mount installation is approved for a maximum weight of more than 150 pounds and a frontal area of 2.5 square feet. - This STC only addresses structural, performance and handling qualities requirements. No system or electrical requirements are specified. Placement of the actual e...</td>
<td>23/05/2012</td>
<td>Active</td>
<td>10039761</td>
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<tr>
<td>AERO PACIFIC</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated/required noise levels and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>23/05/2012</td>
<td>Active</td>
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<td>10039788</td>
<td>TECH-TOOL PLASTICS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC SH875659W</td>
<td>AS 355 N, AS 355 F1, AS 355 E</td>
<td>Installation of aft Cabin Comfort Windows</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>10039789</td>
<td>TECH-TOOL PLASTICS, Inc.</td>
<td>REV. 2</td>
<td>FAA STC SH875659W</td>
<td>AS 350 D, AS 350 BA, AS 350 B1</td>
<td>Installation of aft Cabin Comfort Windows</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10039791</td>
<td>BOMBARDIER INC.</td>
<td>Issue 1</td>
<td>TCCA STC C-LSA12-040/D</td>
<td>BD-700-1A10</td>
<td>Installation, Modifications to Interior Completion - ASN 9028</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10039792</td>
<td>BOMBARDIER INC.</td>
<td>REV. 2</td>
<td>TCCA STC LSA01-329</td>
<td>BD-700-1A10</td>
<td>Seating Limitations for All RHS Divan The TCCA LSTC was revised to allow for Occurrence of Divan Center and/or Seats During Taxi, Take-off and Landing in all Operations</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>STC Holder</td>
<td>Rev. level</td>
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<td>BOMBARDIER INC.</td>
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<td></td>
<td>Installation, Aircraft Completion Modification - ASI 453</td>
<td>Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012; Bombardier Airplane Certification Definition Record RAC-8M700-001 Revision Dated 04 May 2012; Bombardier Airplane Flight Manual Supplement G-FM250S4001 Revision basic instructions for Continued Airworthiness document COA4530000-ICA or later revision.</td>
<td>Restricted to AB 9453 The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment requires a re-examination of the certification basis.</td>
</tr>
<tr>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 4</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td>Installation, Tow Bar Instructions for Continued Airworthiness - MBB/12001</td>
<td>Change Dossier, ref. 11 P05 D200001 Initial issue dated 23rd May 2012.</td>
<td>This change is applicable to aircraft with CPDLC provision enabling Spectracks DLINK with ACARS function previously installed by STC Twenty One prior to installation of this change/repair. It must be determined whether the interchangeability between this change/repair and any other previously installed...</td>
</tr>
<tr>
<td>BOND HELICOPTERS EUROPE Ltd.</td>
<td></td>
<td>EC 635 T2* EC 635 T1 EC 635 P2* EC 135 T2* EC 135 T1 EC 135 P2* EC 135 P1</td>
<td>BOND/M/OO/O06 “Martin- Baker Rotating Seat Installation, EC 135”</td>
<td>The Certification Basis for the original product and the followingadditonal or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement FMS-0500-002, dated May 4th 2012; or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
</tr>
<tr>
<td>FLYING COLOURS CORP.</td>
<td></td>
<td>TCCA STC NO. SA12-33 CL-600-2B19</td>
<td>Installation, All Equipment Bay Tow Bar Stowage Provision</td>
<td>EASA, M.A. 023 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC No. SA12-16 Issue 1 dated 11 April 2012; ACS-IAM-03 Modification Data Summary H477000 Revision NC; Flying Colours Corp. Instructions for Continued Airworthiness for the installation, Tow Bar Stowage Provision, H477011-ICA Revision NC, or later revisions of the above listed documents.</td>
<td>None.</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td></td>
<td>FAA STD SR09137RC-D</td>
<td>EC 135 T2+</td>
<td>Installation of an Extended Right Hand Side Instrument Panel inacconance with SEC Drawing List 135A02-0200. This EASA STC is based on validation of FAA STC SR09137RC-D (Amdt-10-May-2012) see the TIP.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate no/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>None</td>
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<tr>
<td>AEROSPACE DESIGN AND ENGINEERING CO LTD</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td>Modification Compliance List Issue A, dated 30th May 2012 Modification Summary MOD-1083. Issue A, dated 29th May 2012 for modifications. revisions of the above listed documents approved by EASA</td>
<td></td>
</tr>
<tr>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>A330-243</td>
<td>Revise the passenger seat place cabin configuration from FC/24/CL/1893YCL (215 pax) to FC/24/CL/25 pax. passenger placestogether with associated changes to the complete aircraft interior. Installation of 64 passenger places of newly manufactured Sicma Aero - AEROLITE AG</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval. A330 EASA TCDS A. 054 The requirements for environmental protection and the associated certificate no/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td></td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>FAA STC SA09540 AC-D</td>
<td>A96A, C99</td>
<td>Installation of S-TEC system 30 two axis Automatic Flight Guidance Systems (ST-923), in accordance with Master Data ListNo. ST-923-1-0001; Revision A, dated 24-Aug-06 or later FAA approved revision (28V - System).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. Change the certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/ approval.</td>
<td>None</td>
</tr>
<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>210N</td>
<td></td>
<td></td>
<td></td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.</td>
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<tr>
<td>AEROLITE AG</td>
<td>REV. 3</td>
<td></td>
<td></td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate no/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
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<tr>
<td>STC Holder</td>
<td>STC No.</td>
<td>Rev.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>LEES AVIONICS Ltd.</td>
<td>10039536</td>
<td>04/06/2012</td>
<td>Garmin G600 PFND</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable:</td>
<td>Master Data List MDL-1260 Issue 1 dated 31 May 2012; Change Statement: CS-1260 Issue 1 dated 31 May 2012; Flight Manual Supplement: AFMS-1260 Issue 1; or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
</tr>
<tr>
<td>AEROLITE AG</td>
<td>1003944</td>
<td>LBA STC RC 1240</td>
<td>MBB-BK117 C-2</td>
<td>Change to LBA STC No RC 1240 on EMS Equipment P/N 145169-501 This change introduces EMS variant P/N 145169-501 composed by:- Introduction of the new Cargo Platform Inst. P/N 145-0001085-501 - Installation of CIDD Equipment P/N 145169-501 - Removal of Medical...</td>
<td>The Certification Basis for the original product remains applicable:</td>
<td>For EMS Equipment P/N 145170-501, FM-302 (Flight Manual Supplement) For EMS Equipment P/N 145169-501, FM-509 (Flight Manual Supplement) or later revisions of the above listed documents approved by EASA.</td>
</tr>
<tr>
<td>SOFINARE SOCIETEDEFINANCOMEN TOS E REPRESENTACO ES SA</td>
<td>1003948</td>
<td>1900D</td>
<td>INSTALLATION OF TERRAIN AWARENESS AND WARNING SYSTEM (TAWS).</td>
<td>The Certification Basis for the original product remains applicable:</td>
<td>Technical Documents to be Approved by EASA...</td>
<td>None (Applicable to All Models Beechcraft 1900D Airplanes)</td>
</tr>
<tr>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>1003983</td>
<td>DHC-8-202</td>
<td>TCAC STC C-LSA12-006/D</td>
<td>RH Rear Facing Seat Installation, TCAC STC C-LSA12-006/D, Issue 1</td>
<td>FAR 26 as per EASA.M.A.191, Issue 5 (incl. EWS)</td>
<td>The modification is limited to Serial numbers 435, 440 and 441 without-1900 Hearing or C+C+X+Y+Z+ interior only,</td>
</tr>
<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>1004003</td>
<td>REV. 1</td>
<td>B300</td>
<td>Installation of Special Mission antennas and Drop Tube in accordance with Certification documents 2069789 DOC. at issue 3 dated 19 August 2016</td>
<td>The Certification Basis for the original product as amended by the following paragraphs apply:</td>
<td>Approved Aircraft Flight Manual Supplement for Drop Tube Installation Doc. 2069789-AM B, at issue 2, dated 18 July 2016; for later revisions of the above listed documents approved by EASAAirplane Maintenance Manual Supplement Doc. 2069789-AM B, at issue 1 dated 15 July 2016</td>
</tr>
</tbody>
</table>
Installation of Panasonic eXPhone in-flight entertainment system w/In/Phone in accordance with Delta Engineering Master Data List 0107-07606-2280 Rev. E, dated September 13, 2012 or later FAA approved revisions. Validation of FAA STC ST02903NY-D. The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/announcement and remain applicable to this certificate/approval. Aircraft Flight Manual Supplement H1NO-165 Rev. L dated 37.04.2019 for later revisions of the above listed document(s) approved under the EASA system. Product Change Declaration AHSE-6230 Declaration Of Compliance AHSE-6231 Rev. E dated 11.04.2018 up to Revision 4 of this STC – Certification Program AIP.

Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/hazard effect upon the airworthiness of the product.

Installation of LH or RH or both vertical reference antennas installed according to EASA. The below listed required existing systems are not approved by this STC. Revised Supplemental Type Certificate covering the interface with exist in...
<table>
<thead>
<tr>
<th>STC Holder</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>JET AVIATION AG, BASEL</td>
<td>737-800</td>
<td>Indium Antenna Structural Provisions between fuselage station 5S 727C &amp; 737D</td>
<td>IAR 25 Change 13The Certification Basis for the original product remains applicable to this certificate/approval: STC 8330T or STCs: Derogation to latest amendment (Part 21A, 101) Do not materially contribute to safety. The certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/app.</td>
<td>none</td>
<td>26/02/2012</td>
<td>Active</td>
<td>10240150</td>
<td></td>
</tr>
<tr>
<td>BOMBARDIER INC.</td>
<td>BD-700-1A10</td>
<td>Installation. Aircraft Completion Modifications - ASN 9450</td>
<td>EASA.IM.A.009/08The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/app...</td>
<td>Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision K dated 05 June 2012. Bombardier Airplane Flight Manual Supplement G-FM25S04001 Revision BASIC. Instructions for Continued Airworthiness document G09414000-ICA or later revision...</td>
<td>Restricted to ASN 9450 The Wireless Local Area Network (WLAN) is intended to provide internetconnectivity and email services to the airplane's cabin using portableelectronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis.</td>
<td>15/06/2012</td>
<td>Active</td>
<td>10240171</td>
</tr>
<tr>
<td>BOMBARDIER INC.</td>
<td>BD-700-1A10</td>
<td>Installation. Aircraft Completion Modification - ASN 9441</td>
<td>EASA.IM.A.009/08The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/app...</td>
<td>Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012. Bombardier Configuration Definition Record RAZ-BM700-001 Revision K dated 31 May 2012. Bombardier Airplane Flight Manual Supplement G-FM25S04001 Revision BASIC. Instructions for Continued Airworthiness document G09414000-ICA or later revision...</td>
<td>Restricted to ASN 9441 The Wireless Local Area Network (WLAN) is intended to provide internetconnectivity and email services to the airplane's cabin using portableelectronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis.</td>
<td>15/06/2012</td>
<td>Active</td>
<td>10240172</td>
</tr>
<tr>
<td>KUERZI AVIONICS AG</td>
<td>B737-300</td>
<td>GSM mail tracker installation a small GSM device integrates into the aircraft and able to report its current position by means of GSM cell information.</td>
<td>The Certification Basis for the original product and the followingaddition or alternative airworthiness requirements are applicable to this certificate/approval: CRM F-01 (Special Condition/Interpretive Material) Issue 03 &quot;Li-Ion batteries installation&quot; The certificated noise and/or emissions lev...</td>
<td>Project and Order Description POD-201033911 rev. BM Master Document List 201033911 rev. NCInstructions for Continued Airworthiness ICA201033911 rev. NC or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>18/06/2012</td>
<td>Active</td>
<td>10240176</td>
</tr>
<tr>
<td>STANDARD AERO</td>
<td>2, B200C 2, B200 1, 200C 1, 200</td>
<td>Installation of Pratt &amp; Whitney PT6A-52 Engines in accordance with Standard Aero Master Data List document number T016151. Revision &quot;A&quot; dated June 01, 2009 or later FAA approved revision.</td>
<td>The Certification Basis for the original product and the followingaddition or alternative airworthiness requirements are applicable to this certificate/approval as established by FAA STC 902715CH-OT this certificate/approval involves a change to the requirements for environmental protection and...</td>
<td>Airplane Flight Manual Supplement document number 1014604 Revision &quot;A&quot; dated June 05, 2009 or later: Instructions for Continued Airworthiness document number T016175, Revision &quot;A&quot; or later: TCE09 ICA2006 A377 records C10727 through C10730, or later revisions of the above listed documents approved by EASA...</td>
<td>None</td>
<td>15/06/2012</td>
<td>Active</td>
<td>10240181</td>
</tr>
<tr>
<td>INTERNATIONAL ENGINEERING &amp; STANCE (IEA)</td>
<td>F177RG</td>
<td></td>
<td>The Certification Basis for the original product and the followingaddition or alternative airworthiness requirements are applicable to this certificate/approval: CS 23H certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Supplément au manuel de maintenance SUP-ERHC-MANUEL CEMAMENAGEMENT-CESSNA/177RG-2010 Issue D.Supplement au manuel de vol SUP-ERHC-MANUEL DE VOL - CESSNA/177RG Issue G or later revisions of the above listed documents approved by EASA.</td>
<td>Limitation to F177RG serial number 0016.</td>
<td>15/06/2012</td>
<td>Active</td>
<td>10240183</td>
</tr>
<tr>
<td>STC Holder</td>
<td>STC No.</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>AEROSPACE, Inc.</td>
<td>B767-200</td>
<td>B767-200</td>
<td>Installation of VIP Bedroom and Bathroom</td>
<td>EASA IM.A.205 at issue 2 The Certification Basis for the original product and the followeingaddional or alternative airworthiness requirements are applicable to the certificate/approval.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List STC-AT-003, Rev. 3 Supplement to the AFM, AMM-1/STC-AT-044, Rev. 3 Supplement to the AMM, AMM-1/STC-AT-044, Rev. 1 Installation instructions, EASA Master Data List for the listed STC.</td>
<td>None</td>
<td>15/06/2012</td>
</tr>
<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>P200ET</td>
<td></td>
<td>Title: Provisions for radar operation/Description: Installation of a radome below the photogrammetric hatch/activation of the rear hatch (installed with E-OAT-047), installation of the vertical fan, upgraded mission power supply system and removal of hatchdoors and hatch door mechanism (certified with ST...)</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The Certification Basis for the original product remains applicable to this certificate/approval. The Certification Basis for the followeingaddional or alternative airworthiness requirements are applicable to this certificate/approval. The Certification Basis for the followeingaddional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Change Dossier, STC-AT-044, Rev. 3 Supplement to the AFM, AMM-1/STC-AT-044, Rev. 3 Supplement to the AMM, AMM-1/STC-AT-044, Rev. 1 Installation instructions, EASA Master Data List for the listed STC.</td>
<td>STC-AT-003 step 1 or step 3 and EO-AT-047 must be installed prior to thermodesignification.</td>
<td>18/06/2012</td>
<td>Active</td>
</tr>
<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>B747-400</td>
<td>B747-400</td>
<td>Installation of a Panasonic kx2 In-Flight Entertainment System in accordance with Delta Engineering Master Data List 0107-07459-2283 Revision B, dated May 9, 2012 or later FAA approved revisions/Validation of FAA STC ST02901NY-D</td>
<td>The Certification Basis for the original product and the followeingaddional or alternative airworthiness requirements are applicable to the certificate/approval. Section ...Title ...At Amendment25.831 Ventilation , 25.852(5)555 , 25.1168.559 , 25.1168.589 , 25.1168.599.</td>
<td>0107-07459-2283 Rev. B, dated May 9, 2012 Master Data List 0107-07459-2283 Rev. A, dated May 17, 2012 Instructions for Continuation/Airworthiness for Panasonic kx2 IFE system0607-07465-2283 Rev. - Airplane Flight Manual Supplement or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>18/06/2012</td>
<td>Active</td>
</tr>
<tr>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>HGC-6 SERIES 300</td>
<td>HGC-6 SERIES 300</td>
<td>Installation of Universal Avionics CVR-1020AS or KAPTURE Cockpit Voice Recorder with RIPS Option in accordance with Transport Canada STCSA11-80 Issue 2</td>
<td>The Type and OAO Certification Bases (CB) for the original product/airworthiness requirements is applicable to this certificate/approval. The Certification Basis for the alternative product remains applicable to this certificate/approval.</td>
<td>Modification Instructions E103111, Rev. 1 Maintenance Manual Supplement E103131, Rev. D, dated 01/03/2016 Master Data List for the listed STC 0107-07459-2283 Rev. B, dated May 9, 2012 Master Data List 0107-07459-2283 Rev. A, dated May 17, 2012 Instructions for Continuation/Airworthiness for Panasonic kx2 IFE system0607-07465-2283 Rev. - Airplane Flight Manual Supplement or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this change/repair it must be determined that there is no interrelationship between this change/repair and any other previously installed design change and/or repair and any other previously installed design change and/or repair will not introduce unacceptable risks upon the airworthiness of the product.</td>
<td>14/04/2023</td>
<td>Active</td>
</tr>
<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10040249</td>
<td>CESSNA ZURICH CITATION SERVICE CENGBBH</td>
<td>REV. 1</td>
<td>0300, B300C, B200, B200CGT, 8200GT</td>
<td>ADS-B OUT Compliance/Out Dual Collins TDR-840 Transponder Mode S P/N 820-9210-055m, Dual Collins TDR-840 Transponder Mode S P/N 822-9210-055m &amp; 1 datamuseum/switch ADS-B 1/ ADS-B 2 IN HHM</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to aircraft/certification/approval/EOs or ESIFs/CS 23 for the change/TSI certified/noise and/or emissions levels of the original product unchanged and remain applicable.</td>
<td>Master Document List MDL-100 Rev. B 12.06.2012/Instructions for continued airworthiness ICA-100 Rev. A of 02.06.2012/Supplement FMS-100 Rev. 01/22.05.2012</td>
<td>none</td>
<td>11/02/2012</td>
</tr>
<tr>
<td>10040250</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>REV. 1</td>
<td>FAA SR221756</td>
<td>EC635 T2 EC635 T1 (CPDS), EC635 P2 EC635 T2 EC635 T1 (CPDS), EC635 T2 EC635 T1 (CPDS), EC635 P2</td>
<td>Cargo Hook Kit Initial Installation of a Single Cargo Hook Kit for non-human external cargo (HEC) in accordance with Master Drawing List 155-170-00 and FAA STC SR221756 Revisions 1/1 Roduction of Dual Cargo Hook Kit w/PDCS for carrying offshore external cargo (HEC) in accordance with...</td>
<td>The Type and OS2 Certification Bases (CB) for the original product/procurements applicable to this certification/approval. The requirements for environmental protection and the $associated/related/fixed/noise and/or emissions levels of the product are unchanged and remain applicable to this certification/approval w...</td>
<td>For initial STC/Limited to helicopters with CDS/CPDS cockpit variants. Limited to helicopters equipped with cargo hook assembly P/N AS21-8-9. AS22-52-21, or AS21-17-9...</td>
<td>20/12/2022</td>
</tr>
<tr>
<td>10040254</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>FAA SR221756</td>
<td>8275-200</td>
<td>B757-200</td>
<td>Passenger cabin reconfiguration on OpenAirways B757</td>
<td>H4ACP079 - Certification basis/HA48889 - Emergency Evacuation Procedure/AIC-100 - Enhanced Zonal Analysis Procedure/AFM-0002 - Flight Manual Supplement/FMS-100 Rev. 02.05.2012/Revision 03-03-01 - Component Maintenance Manual 220-05-00</td>
<td>None</td>
<td>20/06/2012</td>
</tr>
<tr>
<td>10040256</td>
<td>BOMBARDIER INC.</td>
<td>FAA SR221756</td>
<td>BD-700-1A10</td>
<td>Installation, Aircraft Completion Modifications - AIN 9455.</td>
<td>EASA JM.40009-1The Certification Basis for the original product remains applicable to aircraft/certification/approval. The requirements for environmental protection and the $associated/related/fixed/noise and/or emissions levels of the product are unchanged and remain applicable to this certification/approval w...</td>
<td>Transport Canada STC No. SA12-16 Issue 2 dated 30 March 2012/Bombardier Configuration Definition/Review RAE-JM.40009-1 of 21 June 2012/Bombardier Aircraft Flight Manual Supplement G-FM-221-1-00/AIC-100 - Enhanced Zonal Analysis Procedure/AFM-0002 - Flight Manual Supplement/FMS-100 Rev. 02.05.2012/Revision 03-03-01 - Component Maintenance Manual 220-05-00</td>
<td>None</td>
<td>21/06/2012</td>
</tr>
<tr>
<td>10040257</td>
<td>BRITISH AIRWAYS PLORBRITISH AIRWAYS ENGINEERINGCO. OF AIRWORTHINESS</td>
<td>FAA SR221756</td>
<td>SA12-16</td>
<td>Installation, Aircraft Completion Modifications - AIN 9455.</td>
<td>777-200</td>
<td>Installation: Single or Dual Avionics P/N 804-1205 miniQAR MK III</td>
<td>C525 Amendment 11The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to aircraft/certification/approval. The requirements for environmental protection and the $associated/related/fixed/noise and/or emissions levels of the product are unchanged and remain applicable to this certification/approval w...</td>
<td>Master Data List BA-777-34-003-MLD Issue 00, dated 20 March 2012 or later revisions of the above listed documents approved by EASA</td>
</tr>
<tr>
<td>10040290</td>
<td>CESSNA ZURICH CITATION SERVICE CENGBBH</td>
<td>FAA SR221756</td>
<td>8200, B300C, B200, B200CGT, 8200GT</td>
<td>Mini Quick Access Recorder (QAR) Installation: Single or Dual Avionics P/N 804-1205 miniQAR MK III</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to aircraft/certification/approval. CS-23The associated/noise and/or emissions levels of the original productunchanged and remain applicable to this certification/approval w...</td>
<td>Master Document List MDL-102 Rev. B 12.06.2012/Instructions for Continued airworthiness ICA-100 Rev. A of 02.06.2012/Supplement FMS-100 Rev. 01/22.05.2012</td>
<td>None</td>
<td>22/06/2012</td>
</tr>
</tbody>
</table>
10040294
DARTE AEROSPACE USA, Inc.

STC Holder
Active
System
Active
Date of
NORTHWEST Associated Technical
N0401: Certification of Deployment of In-
DELTA FAA Active
Rev. None
Validation of FAA STC SR02024SE: D119-646 Replacement Skittube Ins.

Certification basis
10040318
STC SR02024SE dated 11 August 2006MDL-
D119-646 Rev. 016A,
D119-646 Rev. 11N-
D119-646 Rev. 0r later revisions of the above listed documents approved by EASA m conformity with EASAED Decision 2004/04/ CF (or subsequent revisions ofthis decision)

Description
10040295
NORTHWEST AEROSPACE TECHNOLOGIES INC.

Installation of Aircraft Completion Modifications - ASN 9437.

STG AEROSPACE Inc.

SAFTGlo, Inc.

installation of provisions for Panasonic eX2 seatback monitor is ins...

STC SR02024SE

Certification basis
STC SR02024SE dated 11 August 2006MDL-
D119-646 Rev. 016A,
D119-646 Rev. 11N-
D119-646 Rev. 0r later revisions of the above listed documents approved by EASA m conformity with EASAED Decision 2004/04/ CF (or subsequent revisions ofthis decision)

10040296
BOMBARDIER INC.

Installation of Aircraft Completion Modifications - ASN 9437.

Certification basis
10040305
DELTA ENGINEERING CORPORATION

Equipment and for the
communications suite (GCS) and the

Certification basis
Transport Canada STC No.
SA12-16 Issue 2 dated 20 March 2012,Bombardier Configuration Definition

10040318
SWS CERTIFICATION SERVICES ltd.

This modification covers:- Installation of Recaro 3410 785 Economy (EC) seats and Airo BusinessClass (BC) seats, fitted with Panasonic RX2 IFE, onto TAP A340-300aircraft. - Installation of AnSafe Airqib System to demonstrate compliance to EASA 5.6.2 when the Panasonic eX2 seatback monitor is ins...

Equipment and for the
communications suite (GCS) and the

Certification basis
- Compliance Report 3091-
COM-01 Issue 1c-<v>-
 dated 12 Oct2012 -
Maintenance Manual 3091-
MMS-01 Issue 1, dated 12 Oct2012 -
Declaration of compliance SW95-C-3091
Issue A, dated 16 Oct 2012 or later revisions of the above listed documentsapproved by EASA

10040330
STG AEROSPACE Inc.ASAFTILO, Inc.

Validation of FAA STC ST03684AT
(Installation of SafTGloPhotoluminescent Escape Path Marking System).

Certification basis
FAA STC ST03684AT, Amendment November 3,
2011STG report AC37-7 Rev. F (Master Drawing
latest amendment FAR Part 25-0
requirements: Regulations at the
alternative airworthiness
to the following additional or
alternative airworthiness requirements: Regulations at the latest amendment FAR Part 25-0 through 25.23(a)-(b); 25.111(a)-(d); 25.113
(a)-(c); 25.121(a)(b),25.123(a)(c); 25.147(a)(f); 25.1...

Certification basis
- These provisions are for a
Panasonic Global
Communications Suite (GCS)
System. The installation and
activation of the hw/GCS system is to be in accordance with a
separate
approval.Additionalapproval is
required for the installation of additional
GCS systemequipment and for the activation of these pr...

Certification basis
Prior to installation of this
modification it must be
determined that the
interrelationship between this
modification and any other
previouslyinstalled modification
and/or repair will introduce no
adverse effectupon the
airworthiness of the product.

Certification basis
none

Certification basis
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Certification basis
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required for the installation of additional
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Certification basis
none
<table>
<thead>
<tr>
<th>STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>10040333</td>
<td>BARCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U.</td>
<td>REV. 1</td>
<td>P 68 OBSERVER 2</td>
<td>P.68 C</td>
<td>&quot;External load mount&quot;, Installation of a mount to hold external loads into existing photogrammetric habitat, according to Certification Plan INAER.M11037/25-PdC, Revision 0, Date: 21/08/2012.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Data Document List INAER.M11037/25-LMD, Revision 0, Date: 21/06/2012 2. Instructions for continued airworthiness INAER.M11037/25-ICA, Revision 0, Date: 21/06/2012 3. Flight manual supplement INAER.M11037/25-FMS, Revision 0, Date: 21/06/2012 or later revisions of the above listed document(s) apply.</td>
<td>None</td>
<td>15/01/2019</td>
<td>Active</td>
<td>10410333</td>
</tr>
<tr>
<td>10040358</td>
<td>VIH AEROSPACE Inc.</td>
<td>REV. 1</td>
<td>TCAC SH08-29</td>
<td>S-92A</td>
<td>LH5 Crew Bubble Window Installation Revision 1 is to add limitations for flight into known icing conditions for night time operations per TCAC STC issue 5 dated October 05, 2018.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>VS92-0803-MDL Rev. 7 dated Jan 10, 2014 -- Master Document List/VS92-0803-MDL Rev. 3 dated Sep 17, 2018 -- Flight Manual Supplement/VS92-0803-ICA Rev. 4 dated Jan 10, 2014 -- Instructions for Continued Airworthiness -- later revisions of the above listed document(s) approved/accepted on behalf of EASA in...</td>
<td>Flight into known or forecasted icing conditions not approved with the Bubble Windows installed. Prior to installation of this change/repair it must be determined that there is no adverse effect...</td>
<td>29/05/2020</td>
<td>Active</td>
<td>10410358</td>
</tr>
<tr>
<td>10040394</td>
<td>AERODATA AG</td>
<td></td>
<td>B300</td>
<td></td>
<td>Installation of Aerodata Automatic Flight Inspection System 4D-AFIS-0280</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. CS-23, Amendment 2: The requirements for environmental protection and the associated certified noise and/or emissions levels of the aircraft remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Flight Manual Supplement: AFM-121, Rev. 2 or later revisions of the above listed documents approved by EASA -- Installation Manual HB-A/-C-1090, Rev. 0 -- Master Document List MDL-A/-C-0170, Rev. 2: Maintenance Manual Supplement AMM-1125, Rev. 0.</td>
<td>none</td>
<td>28/06/2012</td>
<td>Active</td>
<td>10410394</td>
</tr>
<tr>
<td>10040380</td>
<td>ALTRAN LAB/a AICAO</td>
<td></td>
<td>F900EX</td>
<td></td>
<td>Partial Cabin Entertainment and Galley equipment upgrade</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>File ref.: 12006 Certification Program CP-12006, Revision A, dated 27/06/2012, Service Bulletin ACDS-SB-12006, Revision A, dated 27/06/2012, or later revisions of the above listed documents approved by EASA</td>
<td>Falcon 910EX S/N 054</td>
<td>28/06/2012</td>
<td>Active</td>
<td>10410380</td>
</tr>
<tr>
<td>10040394</td>
<td>HELI-ONE (NORWAY) AS</td>
<td></td>
<td>AS 332 L, AS 332 L1</td>
<td></td>
<td>Goodrich Dual Bond Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable.</td>
<td>AFM Supplement AHS-009, dated 07.04.2011 -- Continued Airworthiness Supplement No.: AHSE-55211, Rev. B, dated 12.01.2012, or later revisions of the above listed documents approved by EASA -- Master Drawing AHSE-55200, Rev. B, dated 20.06.2011.</td>
<td>None</td>
<td>29/06/2012</td>
<td>Active</td>
<td>10410394</td>
</tr>
<tr>
<td>10040405</td>
<td>LEARJET Inc.</td>
<td></td>
<td>FAA STC SAB192NM-D</td>
<td></td>
<td>Installation of a Sextant Integrated Electronic Standby Instrument System</td>
<td>EASA IA A 2009 The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. SAB192NM-D dated 02 August 2001 -- Learjet Master Document List GCC3423501, Revision G, dated 21 January 2004.</td>
<td>none</td>
<td>02/07/2012</td>
<td>Active</td>
<td>10410405</td>
</tr>
</tbody>
</table>
### List of EASA Supplemental Type Certificates

<table>
<thead>
<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Certification basis</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>10040411</td>
<td>AAR AIRCRAFT SERVICES</td>
<td></td>
<td>FAA STC ST03190CH</td>
<td>B717-200</td>
<td>Interior Reconfiguration</td>
<td>EASA MA 211. Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. ST03190CH dated 25 June 2012</td>
<td>None</td>
<td>02/07/2012</td>
<td>Active</td>
<td>10040411</td>
<td></td>
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<tr>
<td>10040427</td>
<td>AIM AVIATION (HENSHALLS) LTD</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B747-400</td>
<td>Installation of galley 2A, 2B, 2C and 2D and cabinets in the 4A complex in a Boeing 747-400 to be operated by Virgin Atlantic Airways.</td>
<td>The Type and OSD Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>AIM Henshalls STC Definition Document dated 21 June 2011.</td>
<td>None</td>
<td>02/07/2012</td>
<td>Surnanded</td>
<td>10040427</td>
<td></td>
<td></td>
</tr>
<tr>
<td>10040440</td>
<td>ONSBOARD SYSTEMS INTERNATIONAL</td>
<td>FAA STC SR01684SE</td>
<td>BO 105 S, BO 105 LS A-3</td>
<td>Validation of FAA STC SR01684SE - Cargo Hook Kits</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA approved Onboard Systems Rotorcraft Flight Manual Supplement RFMS121-029-00, dated March 1, 2012.</td>
<td>None</td>
<td>03/07/2012</td>
<td>Active</td>
<td>10040440</td>
<td></td>
<td></td>
</tr>
<tr>
<td>10040450</td>
<td>MARSHALL OF CAMBRIDGE AEROSPACELIMI TED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A340-312</td>
<td>This modification intends to improve the airflow.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- FAA approved Onboard Systems Master Data List ERP 11-062 Revision C dated 23 June 2012 for Continued Airworthiness (ICA). AAR Aircraft Services Report No. ERP 11-066 Revision IR, dated 28 March 2012 or later revisions of the above listed documents app...</td>
<td>None</td>
<td>25/10/2012</td>
<td>Active</td>
<td>10040450</td>
<td></td>
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</table>

**EASA STC No.**: EASA Supplemental Type Certificate Number.

**STC Holder**: The company or organization that holds the Supplemental Type Certificate.

**Rev. level**: Revision level of the Supplemental Type Certificate.

**Foreign/NAA STC ref.**: Reference to the foreign or national Supplemental Type Certificate.

**Models**: Models covered by the Supplemental Type Certificate.

**Certification basis**: The basis for the certification.

**Description**: A description of the modification or change.

**Certification basis**: The basis for the certification.

**Associated Technical Documentation**: The associated technical documentation.

**Limitations/Conditions**: Limitations and conditions associated with the modification or change.

**Date of Issue**: The date the Supplemental Type Certificate was issued.

**Status**: The status of the Supplemental Type Certificate (Active or Surnanded).

**System Record**: The system record number associated with the Supplemental Type Certificate.
<table>
<thead>
<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
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<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>10040456</td>
<td>AEROTECH &amp; CONCEPT</td>
<td></td>
<td></td>
<td>EC225LP</td>
<td>SYSTEME ANTI-GRATOREStabilized load stabilization system</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to both certificates/approval CS.29 Issue 2 paragraphs 301, 303, 305, 307, 337, 663, 610, 666, 1581, 1353, 1357, 1301, 1309, 1322, 1505, 1523, 1528, 1541, 1581The requi...</td>
<td>Flight Manual Supplement EC 225 LP MOD. AEROTECH CONCEPT No. 1087'SYSTEME ANTI-GRATORE; document AE1078BM rev 0 dated02/05/2012or later revisions of the above listed documents approved by EASA- INSTRUCTIONS FOR CONTINUED AIRWORTHINESS NOO 1078 EC225 LP FOR RTE;ifth SYSTEME ANTI-GRATORE BUVI D.</td>
<td>04/07/2012</td>
<td>Active</td>
<td>10040456</td>
<td></td>
</tr>
<tr>
<td>10040468</td>
<td>AVIONIK STRAUBING ENTwicklungs GmbH</td>
<td></td>
<td>PA-42, PA-42-720, PA-42-1000</td>
<td>Installation of Garmin G9600 System with Synthetic Vision (SVT)</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to both certificates/approval CS.23, Amendment2 (not required by 21A.101)The requirements for environmental protection and the associated certificated noise and or emi...</td>
<td>Flight Manual Supplement G9600: ASR-2012-016- FMS-05-00, Rev.000later revisions of the above listed documents approvedby EASA- Instructions for Installation: ASR-2012-016- IFI-10-00, Rev.00- Instructions for Continued Airworthiness: ASR-2012-016- ICA-11-00, Rev.00</td>
<td>none</td>
<td>05/07/2012</td>
<td>Active</td>
<td>10040468</td>
<td></td>
</tr>
<tr>
<td>10040470</td>
<td>ARMSTRONG AEROSPACE Inc.</td>
<td>FAA STC</td>
<td>767-300</td>
<td>Installation of Board Connect InFlight Entertainment System</td>
<td>The Certification Basis for the original product remains applicable to both certificates. The requirements for environmental protection and the associated certificated noise and or emissions levels of the original product are unchanged and remain applicable to this approval.</td>
<td>Master Drawing List DC104-9004-00, Revision B, 22/02/2012AFM Supplement DC363-9004-00, Revision -, 16/02/2012or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2000/04/CF (or subsequent revisions or this decision)</td>
<td>This system is intended to provide internet connection and e-mail services using portable electronic devices (PEDs). Any other intended function of this equipment will require further certification approval.</td>
<td>05/07/2012</td>
<td>Active</td>
<td>10040470</td>
<td></td>
</tr>
<tr>
<td>10040472</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>MU-300 (DIAMOND IA) MU-300 (DIAMOND I)</td>
<td>Digital DOR Installation</td>
<td>The Certification Basis of the original product and the following additional or alternative airworthiness requirements are applicable to both certificates. For the change CS.25, Amdt. 4 has been elected to comply with where possible. The certificated noise and or emissions levels of the original product are unchanged and remain applicable to this approval.</td>
<td>Installation in accordance with EASA approved Engineering Order, Doc.No. 2076242-EO, Issue 1, dated 26-June-2012, or later approved issue Operation in accordance with EASA approved Aircraft Flight Manual Supplement, Doc. No. 2076242-AM, Issue 1, dated 25-June-2012 (&lt;&gt;) Date, &lt;&gt; or later approved issue</td>
<td>This change is limited to the configuration of aircraft/equipment types listed under Chapter 4 in EASA approved Engineering Order, Doc. No. 2076242-EO, Issue 1, dated 26-June-2012, or later approved issue.</td>
<td>09/07/2012</td>
<td>Active</td>
<td>10040472</td>
<td></td>
</tr>
<tr>
<td>10040478</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC NO. SA12-16</td>
<td>BD-700-1A11</td>
<td>Installation, Aircraft Completion Modifications - ASN 9457</td>
<td>EASA.31.A.009</td>
<td>Transport Canada STC No. SA12-16 Issue 2, dated 29 March 2012 Bombardier Configuration Definition Record RAC-BMT071-001 Revision H dated 28 May 2012 Bombardier Aircraft Flight Manual Supplement G-FMD056K01 RevisionBASIC.Instructions for Continued Airworthiness Document G04570000- ICA or later revis...</td>
<td>Restricted to ASN 9457 The Wireless Local Area Network (WLAN) is intended to provide internetconnectivity and email services to the airplane’s cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require examination of the certification basis.</td>
<td>05/07/2012</td>
<td>Active</td>
<td>10040478</td>
<td></td>
</tr>
<tr>
<td>STC Holders</td>
<td>STC no.</td>
<td>Rev.</td>
<td>Foreign/NAA STC no.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>CAA UK RESPONSIBIL ITY</td>
<td>10040493</td>
<td>CL-600-2A12</td>
<td>Introduction of Swift Broadband</td>
<td>EASA.M.0237</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. STC TwentyOne Ltd Master Document No. S21.MDL-079 Issue 3. Flight Manual Supplement No. S21.FMS-078 issue 1 or later revisions of the above listed documents approved by EASA.</td>
<td>none</td>
<td>06/07/2012</td>
<td>Active</td>
<td>10040493</td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

| AAR CARGO SYSTEMS | 10040510 | REV. 1 | FAA STC ST09658R | A300B4-622, A300B4-622 R, A300B4-605 R, A300B4-620, A300B4-601 | Validation of FAA Supplemental Type Certificate (STC) ST09658R (Install a Main Deck Cargo Loading System (AAR Part Number 58535-101) on Airbus A300-600 series aeroplanes in accordance with AAR Master Drawing List Report Number 2149 Revision B, dated 15 December 2008). | The Certification Basis for the original product remains applicable to this certificate/approval. | AAR Weight & Balance Report A300-600 FSI Main Deck Cargo Loading System (MDCLS) P/N 58535-101 No. 2147 Rev A, dated 25 Aug. 2008 AAR Instructions For Continued Airworthiness A300-600 FSI MDCLS P/N 58535-101 No. 2148 original issue, dated 16 July 2007 or later revisions of the above listed documents approved by EASA. | none | 13/07/2012 | Active | 10040510 |

| 39 CERTIFICATION, LLC | 10040513 | REV. 2 | FAA STC ST0347CH | 747-400 | Certification of Compatibility with Cellular and Wireless Data Technologies. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at the later amendment 25.1309 of CS-25 administration 2 and 25.1529 of CS-25 at amendment 2. The requirements for environmental protection and... | Armstrong Aerospace Master Data List DC104-9028-00, Revision G, dated 19 January 2013 Armstrong Aerospace Aircraft Equipment Compatibility List C3307-9028-01, Revision B, dated 18 January 2013 Armstrong Aerospace Aircraft Equipment Compatibility List C3307-9028-02, Revision B, dated 18 January 2013 | The following are the limitations associated with this typeapproval: 1. This STC does not constitute installation approval of aeroplane hardware. It only provides data that may partially support their installation or activations. This STC may only be applied to an aeroplane when an airplane installation... | 05/08/2014 | Active | 10040513 |

| INNOTECH AVIATIONA DIVISION OF IMP GROUP LTD | 10040515 | TCCA STC NO. Q. LSA07-040D | BD-700-1A10 | Installation of Iridium Phone System and Interior modification - ASN9028 | EASA.M.0099 | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Transport Canada STC No. Q.LSA07-040D Issue 3, dated 07 June 2011 Innotech Aviation MDC-21173-002 Master Document List - Installation of Iridium Phone System and Minor Interior Modifications on ACMSN 9628 Revision B, dated 07 June 2011 Innotech Aviation TL2172-004, Time Limits and Maintenance Checks... | Restricted to ASN 9028 Previous installation of EASA STC No. 10039790 and EASA STC No. 10039791 fleet pre-requirements. | 06/07/2012 | Active | 10040515 |

| ACC COLUMBIA Jet Service GmbH | 10040526 | REV. 1 | PA-31T | PA-31T Camera Noise Modification/Change Note CN_2005593_rev0_20120201Rev. 1 Transfer to ACC Columbia Jet Service GmbH | The Certification Basis for the original product remains applicable to this certificate/approval. Approval FAR 23-Amend 17: The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | MDOCL_2025593_rev0_20120201Installation Instructions, dated 07 January 2013 | as referred in AFM-S | 26/01/2022 | Active | 10040526 |

| H4 AEROSPACE (UK) Ltd. | 10040531 | CAA UK RESPONSIBIL ITY | 767-200, -300 | Introduction of an AD Aerospace Cockpit Door and Cabin Surveillance System on Boeing 767-200/300. | EASA.M.0355 | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | H4 Aerospace Modification Summary and Declaration of Compliance No. MD4HA1060 Issue 3, dated 06 July 2012 or later revision of the above listed documents approved by EASA. | none | 09/07/2012 | Active | 10040531 |
List of EASA Supplemental Type Certificates

<table>
<thead>
<tr>
<th>STC Holder</th>
<th>Status</th>
<th>STC no.</th>
<th>Rev.</th>
<th>Link to STC</th>
<th>Description</th>
<th>Certification basis</th>
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<tbody>
<tr>
<td>AIR METHODS CORPORATION</td>
<td>Active</td>
<td>10040541</td>
<td>FAA STC</td>
<td>SR00523DE</td>
<td>Installation of Emergency Medical Interior</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>-080/0265 revision A dated 18 May 2012, Flight Manual Supplement later revisions of the above listed documents approved by EASA in accordance with EASA-ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>09/07/2012</td>
<td>None</td>
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<tr>
<td>AIR METHODS CORPORATION</td>
<td>Active</td>
<td>10040543</td>
<td>FAA STC</td>
<td>SR00522DE</td>
<td>Installation of two (2) aft jump seats</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>-080/0265 revision A dated 18 May 2012, Flight Manual Supplement later revisions of the above listed documents approved by EASA in accordance with EASA-ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>09/07/2012</td>
<td>None</td>
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<tr>
<td>AIR METHODS CORPORATION</td>
<td>Active</td>
<td>10040545</td>
<td>FAA STC</td>
<td>SR00524DE</td>
<td>Installation of Articulating Patient Loading System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>-080/0265 revision A dated 18 May 2012, Flight Manual Supplement later revisions of the above listed documents approved by EASA in accordance with EASA-ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>09/07/2012</td>
<td>None</td>
<td>10040545</td>
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<tr>
<td>ALEISCO CORPORATION A/S</td>
<td>Active</td>
<td>10040578</td>
<td>REV. 1</td>
<td>209B</td>
<td>Installation of Video Transmission Unit in accordance with Master DataList PP-008-MDL-04, issue 4, dated 11 July 2012 or later approved revision.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>-080/0265 revision A dated 18 May 2012, Flight Manual Supplement later revisions of the above listed documents approved by EASA in accordance with EASA-ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>21/01/2013</td>
<td>Installer must check availability of EASA-approval for one or more referenced STC when airplane is modified per that STC(s).</td>
<td>10040578</td>
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<tr>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>Active</td>
<td>10040583</td>
<td>A300F4-606ST</td>
<td>Special Transport - Instrumented Airbus Sectional - 1 of 35 - Modification STS50000000 (applicant’s reference)</td>
<td>Installation of Video Transmission Unit in accordance with Master DataList PP-008-MDL-04, issue 4, dated 11 July 2012 or later approved revision.</td>
<td>-080/0265 revision A dated 18 May 2012, Flight Manual Supplement later revisions of the above listed documents approved by EASA in accordance with EASA-ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>06/12/2021</td>
<td>None</td>
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<td>STC No.</td>
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<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10040851</td>
<td>S-TEC CORPORATION</td>
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<td>FAA STC: SR002346L A</td>
<td>BELL 206L-3, 206L-4 BELL 206B, 206L-1, 206L-2</td>
<td>Installation of HAMSAS stability augmentation system and autopilot validation of FAA STC SR002346L A</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this STC certificate/approval. Far 27, with following amendments and related paragraphs: Amdt 11: 1222, Amdt 13: 1321, 1351, 1357, Amdt 14: 29, 1581, Amdt 18: 1520, Amdt: ...</td>
<td>VFR only</td>
<td>17/01/2012</td>
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<tr>
<td>10040866</td>
<td>AIRCRAFT STRUCTURES INTERNATIONAL CORP.</td>
<td></td>
<td>FAA STC: SA10840SC</td>
<td>CESSNA 208, CESSNA 208B</td>
<td>Installation of Jack Pads on Cessna 208 Series aircraft in accordance with FAA STC SA10840SC</td>
<td>The Certification Basis for the original product remains applicable to this STC certificate/approval. The requirements for environmental protection and the associated/certified noise/ emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Assistance STC Manual Supplement, Hel/SAS- RFMS-206 revision N Coated March 1, 2011-Master Data List, Hel/SAS- MDA-2068 revision E dated March 19, 2012 Instructions for Continued Airworthiness (ICA), Document No. AA-05021Revision B dated November 23, 2011 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/014/CF (or subsequent revisions of this decision).</td>
<td>None</td>
<td>11/07/2012</td>
<td>Active</td>
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<tr>
<td>10040869</td>
<td>LUFTHANSA TECHNIS AG</td>
<td></td>
<td>B747-8</td>
<td>Camera installation in the vertical stabilizer for ground maneuvering service.</td>
<td>The Certification Basis for the original product remains applicable to this STC certificate/approval. The requirements for environmental protection and the associated/certified noise/emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Assistance STC Manual Supplement, Hel/SAS- RFMS-206 revision N Coated March 1, 2011-Master Data List, Hel/SAS- MDA-2068 revision E dated March 19, 2012 Instructions for Continued Airworthiness (ICA), Document No. AA-05021Revision B dated November 23, 2011 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/014/CF (or subsequent revisions of this decision).</td>
<td>None</td>
<td>12/07/2012</td>
<td>Active</td>
<td>10040598</td>
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<tr>
<td>10040920</td>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td>A330-223 / -233 A330-201 / -202 / -203</td>
<td>Panasonic eXPhone System Activation on A330 with D2E IF and xConnect. This STC installs a Panasonic eXphone System allowing, in conjunction with other devices, the use of cellphones during flight.</td>
<td>The Certification Basis for the original product remains applicable to this STC certificate/approval. The requirements for environmental protection and the associated/certified noise/emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, Ref : 0496-03-A-2103-R00, dated 11 July 2012Instruction for Continued Airworthiness, Ref : 0496-03-A-2211-R00, dated10 July 2012Aircraft Flight Manual Supplement, Ref : 0496-03-A-2301-R00, dated 10 July 2012Wright and Balance Report, Ref. 0496-03-A-3401-R00, dated 10 July 2012</td>
<td>None</td>
<td>12/07/2012</td>
<td>Active</td>
<td>10040592</td>
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<td>10040901</td>
<td>BOMBARDIER INC.</td>
<td></td>
<td>TCIA STC: SA12-16 ISSUE 2</td>
<td>BD-700-1A11</td>
<td>Installation, Aircraft Certification Modifications - ASN 9461</td>
<td>The Certification Basis for the original product remains applicable to this STC certificate/approval. The requirements for environmental protection and the associated/certified noise/emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC No. SA12-16 Issue 2, dated 20 March 2012Bombardier Configuration Definition Record RAS-SM700-001 Revision M dated 22 June 2013Bombardier Aircraft Flight Manual Supplement G-PAS0554001 Revision BASONIstructions for Continued Airworthiness Document GCBH010000- ICA or later revision.</td>
<td>Restricted to ASN 9461The Wireless Local Area Network (WLAN) is intended to provide internetconnectivity and email services to the airplane’s cabin portable/electronic devices (PEDs). Any other intended function of this equipment will require an examination of the certification basis.</td>
<td>12/07/2012</td>
<td>Active</td>
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<tr>
<td>10040902</td>
<td>PMV ENGINEERING</td>
<td></td>
<td>REV 1</td>
<td>A330-200, A330-300</td>
<td>Installation of provisions for the use of EFB Class 2 on Airbus A330. This modification deals with the installation of a dedicated Electronic Flight Bag (EFB) EIC Esterline Pilot View system on Airbus A330-200/300. The modification is defined as follows: Installation of the Arm Mounts in the cockpit.</td>
<td>The Certification Basis for the original product remains applicable to this STC certificate/approval. A330 EASA TCDS A 004401 whereupon for environmental protection and the associated/certified noise/emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>- Certification Investigation Report AVI-0034-CIR-R04 Issue 6, dated 07/02/2012 for Continued Airworthiness AVI-0034-ICA-R02 Issue 2 dated 25/06/2012Aircraft Flight Manual Supplement AAPSS AVI-0034-AMPS-R02 Issue 2, dated 02/02/2012Airworthiness Maintenance Manual Supplement AMMS AVI-0034-AMMS</td>
<td>None</td>
<td>13/07/2012</td>
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<td>Associated Technical Documentation</td>
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<td>10040616</td>
<td>ASI INNOVATION</td>
<td>10040624</td>
<td>2, 890, E90</td>
<td>1, C90, C90A, C90GT, C90GT</td>
<td></td>
<td>VHF, UHF and GPS antennas installation provisions according to ASI Modification no ADI-FAAM-0066-00</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Flight Manual Supplement no ADI-FAAM-0066, original issue dated 20 June 2012.</td>
<td>13/07/2012</td>
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<td>10040624</td>
<td>ALTRAN LAB/a</td>
<td>10040624</td>
<td>F2000</td>
<td>ST01376W-K-D</td>
<td></td>
<td>Installation of a sliding door, main entryway door removable, galley shelves, and reassembly</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Certification Program CP-12001 Revision A, dated 13 July 2012. Service Bulletin ACD-SB-12001 Revision B or later revisions of the above listed documents approved by EASA.</td>
<td>Falcon 2000 S/N 111</td>
<td>19/07/2012</td>
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<td>10040625</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>10040625</td>
<td>A321-231</td>
<td>ST01376W-K-D</td>
<td></td>
<td>Introduction of Panasonic MPES IFE systemrev1: Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Certification Program CP-12001 Revision A, dated 13 July 2012. Service Bulletin ACD-SB-12001 Revision B or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>27/04/2021</td>
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<td>10040627</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10040627</td>
<td>10040627</td>
<td>ST00803LA-D</td>
<td></td>
<td>Installation of Satcom Antenna (Racial).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>None</td>
<td>16/07/2012</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<td>Status</td>
<td>System Record</td>
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<td>LUTTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>FAA STC NO. SAT178SW-D</td>
<td>T210K</td>
<td>Installation of S-TEC manual Electric trim system. (ST-439), according to Subpart 533, dated 11-May-1988 and Master Drawing List No. 92474, dated 11-May-1988 or later FAA approved revision (14V - System).</td>
<td>Design Non-significant change The Certification Basis of the theoretical product remains unchanged.</td>
<td>FAA approved S-TEC Master Drawing List No. 92474 dated 11-May-1988 or later FAA approved revision.</td>
<td>T210K is eligible also when modified per STC SA2689SW (RAIM modification). Installer must check availability of FAA-approved revision for one or more relevant STC and/or for the airplane modified per that STC(s). If the holder agrees to permit another person to use this certificate to alter the product, the holder states that the product remains unchanged.</td>
<td>17/07/2012</td>
<td>Active</td>
<td>10040647</td>
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<td>WESTERN AVIONICS</td>
<td>REV. 1</td>
<td>FAA STC NO. SAT178SW-D</td>
<td>T209K</td>
<td>Installation of S-TEC manual Electric trim system. (ST-439), according to Subpart 533, dated 11-May-1988 and Master Drawing List No. 92474, dated 11-May-1988 or later FAA approved revision (14V - System).</td>
<td>Design Non-significant change The Certification Basis of the theoretical product remains unchanged.</td>
<td>FAA approved S-TEC Master Drawing List No. 92474 dated 11-May-1988 or later FAA approved revision.</td>
<td>T210K is eligible also when modified per STC SA2689SW (RAIM modification). Installer must check availability of FAA-approved revision for one or more relevant STC and/or for the airplane modified per that STC(s). If the holder agrees to permit another person to use this certificate to alter the product, the holder states that the product remains unchanged.</td>
<td>17/07/2012</td>
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<td>AEROLITE AG</td>
<td>REV. 2</td>
<td>FAA STC NO. SAT178SW-D</td>
<td>T210K</td>
<td>Installation of S-TEC manual Electric trim system. (ST-439), according to Subpart 533, dated 11-May-1988 and Master Drawing List No. 92474, dated 11-May-1988 or later FAA approved revision (14V - System).</td>
<td>Design Non-significant change The Certification Basis of the theoretical product remains unchanged.</td>
<td>FAA approved S-TEC Master Drawing List No. 92474 dated 11-May-1988 or later FAA approved revision.</td>
<td>T210K is eligible also when modified per STC SA2689SW (RAIM modification). Installer must check availability of FAA-approved revision for one or more relevant STC and/or for the airplane modified per that STC(s). If the holder agrees to permit another person to use this certificate to alter the product, the holder states that the product remains unchanged.</td>
<td>17/07/2012</td>
<td>Active</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>LUTTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>FAA STC NO. SAT178SW-D</td>
<td>T210K</td>
<td>Installation of S-TEC manual Electric trim system. (ST-439), according to Subpart 533, dated 11-May-1988 and Master Drawing List No. 92474, dated 11-May-1988 or later FAA approved revision (14V - System).</td>
<td>Design Non-significant change The Certification Basis of the theoretical product remains unchanged.</td>
<td>FAA approved S-TEC Master Drawing List No. 92474 dated 11-May-1988 or later FAA approved revision.</td>
<td>T210K is eligible also when modified per STC SA2689SW (RAIM modification). Installer must check availability of FAA-approved revision for one or more relevant STC and/or for the airplane modified per that STC(s). If the holder agrees to permit another person to use this certificate to alter the product, the holder states that the product remains unchanged.</td>
<td>17/07/2012</td>
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<td>AERODISA S.L.</td>
<td>B300</td>
<td>FMS Upgrade for King Air B300.</td>
<td>Technical Data to be Approved by EASA/AERODISA Master Data List AED2011-17M_MDL ED Rev 04or later revisions of the above listed documents approved by EASA.</td>
<td>S/N Limitation AIRPLANE MODEL HAWKER BEECHCRAFT B300 S/N FL-255</td>
<td>16/07/2012</td>
<td>Active</td>
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<tr>
<td>SOCATA</td>
<td>DHC-6 SERIES 300</td>
<td>Avionics upgrade based on Garmin G600 EFIS and GNI 430 W system, according to SOCATA modification file n° STC. 06.001.11</td>
<td>The Certification Basis for the original product remains applicable toots certificated approval.</td>
<td>Socata Flight Manual Supplement No.3B. Issue 1 dated 15 March 2012. Socata Maintenance Manual Supplement revision 6 dated 15 February 2012. Socata Modification Sheet n° STC. 001001.11. Edition 1 dated 4 July 2012 or later revisions of the abovedicted documents approved by EASA.</td>
<td>17/07/2012</td>
<td>Active</td>
<td>10040659</td>
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<tr>
<td>BRISTOW HELICOPTERS LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>Activation of ACS-B Function using Mode S Extended Spooler</td>
<td>The Certification Basis for the original product remains applicable toots certificated approval. CS29 Amendment 2. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the origin...</td>
<td>Bristow Helicopters Limited Sikorsky SH11 Flight Manual Supplement No.30/Initial Issuer later revisions of the above listed documents approved by EASA/Modification MODI S81/34.5719 Issue 1 dated 23 May 2012 Certification Plan BHL/S61/5607 Initial Issue dated 23 March 2012</td>
<td>none</td>
<td>18/07/2012</td>
<td>Active</td>
<td>10040676</td>
<td></td>
<td></td>
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<tr>
<td>PFW ENGINEERING DO GmbH</td>
<td>747-8</td>
<td>Structural provisions for 747-8 belly camera installation between the NLG bay and the belly fairing at the lowest point of the fuselage.</td>
<td>The Certification Basis for the original product remains applicable toots certificated approval.</td>
<td>STC Description Document DE-530-B747A Issue 01(=Master/Direct List CCN-B747A-MDL Issue 01) or STC Operations Manual B747A-OPM-A dated 31 Jan 2012 or later revision of the above listed documents approved by EASA.</td>
<td>none</td>
<td>11/10/2013</td>
<td>Active</td>
<td>10040680</td>
<td></td>
<td></td>
</tr>
<tr>
<td>AEROFDC WA LP</td>
<td>AS350 B, AS350 B1, AS350 B2, AS350 B3, AS350 BA</td>
<td>Revision to EASA STC 10040685, rev. 1 Installation of an engine air filter in accordance with FDC Master Data List No. 1350-1001 and FDC Installation Instructions 1350-INST-Notes.1. This STC was initially issued to INTERHELICOPTERS, Inc. on 25 October 2004 under reference EASA.IM.R.S.00273 and was later</td>
<td>The Certification Basis for the original product remains applicable toots certificated approval.</td>
<td>Design and Installation Documentation FDC Master Data List No. 1350-1001 revision H dated 20 May, 2011FDC Installation Instructions 1350-INST-1, revision H dated 20 May, 2011Maintenance documentation FDC Instructions for Continued Airworthiness 1350- SERIES-SCA-1, revision E dated April 20, 2011or later rev...</td>
<td>none</td>
<td>31/01/2014</td>
<td>Active</td>
<td>10040685</td>
<td></td>
<td></td>
</tr>
<tr>
<td>ACS-NAI Ltd.</td>
<td>DHC-8-301,.311,.314,.315 DHC-8-201,.202, DHC-8-102,.103,.106</td>
<td>Installation of an improved AR Jet Pipe Support TCCA STC. N 5A01-17, Issue 3</td>
<td>FAR 25 as per EASA, M.A.191 Issue 5 incl. EIWS/The Certification Basis for the original product remains applicable toots certificated approval.</td>
<td>Installation Must be in accordance with Aviation Consulting Service Ltd. Modification/Data Summary N° C 54650, Revision B.Maintenance Must be in accordance with Aviation Consulting Service Ltd. Maintenance/Manual Supplement N° C 54672, N/C or later revisions of the above listed documents approved by EASA inac...</td>
<td>None.</td>
<td>19/07/2012</td>
<td>Active</td>
<td>10040686</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>STC Hold No.</td>
<td>Models</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>2 EXCEL DESIGN LIMITED</td>
<td>REV.3</td>
<td>STC 0038561</td>
<td>PA-31</td>
<td>Provisions for the installation of a Sensor Tunnel</td>
<td>The Certification Basis for the original product remains applicable to the STC/Approval. The requirements for environmental protection and the associated certification remain unchanged and remain applicable to this certificate/Approval.</td>
<td>- Master Document List S21.MDL-1027 Issue 6 - Flight Manual Supplement S21.FMS-0091 Issue 4 - Compliance Record S21.COMP-1027 Issue 4 - Instructions for Continuing Airworthiness S21.TEC-0374 Issue 2 or later revisions of the above listed documents approved by EASA</td>
<td>With a sensor turret with a diameter greater than 15.5&quot; installed flight into known or forecast icing conditions is prohibited. Limitations as defined in AFMS S21.FMS-0091 Issue 4 - The installation and subsequent operation of a sensor turret requires separate STC approval. Prior to installation of this mod...</td>
<td>25/02/2017</td>
<td>Active</td>
<td>10040501</td>
</tr>
<tr>
<td>TENENCIA Ltd.</td>
<td></td>
<td>STC 0038571</td>
<td>BAE 146 SERIES 300 A319-115</td>
<td>ADS-B Out Surveillance</td>
<td>The Certification Basis for the original product remains applicable to the STC/Approval. The requirements for environmental protection and the associated certification remain unchanged and remain applicable to this certificate/Approval.</td>
<td>Modification TEN 016 Issue 10 dated 13 June 2012 Flight Manual Supplement TEN/16/016 Issue 10, or later revisions of the above listed documents approved by EASA</td>
<td>Configuration using transponder part number 753-7800-11005 does not meet EASA AMC 20-24 for ADS-B NAA. Emergency Status/Emergency Indicator arent provided.</td>
<td>20/07/2012</td>
<td>Active</td>
<td>10040730</td>
</tr>
<tr>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV.1</td>
<td>STC 0038581</td>
<td>A321-231</td>
<td>Introduction of Twin Class Cabin Interior Configuration Rev 1 - Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the STC/Approval.</td>
<td>- Modification Summary ref. Master Doc. AES-321-2059, Issue 2 dated 19th July 2012 - Master Data List ref. doc. No. AES-321-2059MDL, Issue 2, dated 19th July 2012 - Project Specific Certification Plan (PSCP) ref. doc. No. AES-TR-054A, Issue 3, dated 19th July 2012, Certification Compliance Report...</td>
<td>The EASA STC 10038863 is a prerequisite for the installation of this STC.</td>
<td>27/04/2021</td>
<td>Active</td>
<td>10040704</td>
</tr>
<tr>
<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td></td>
<td>STC 0038591</td>
<td>B200C</td>
<td>Removing of the metal parts of a cargo pod and provisions for the installation of signal transmission antennas inside the pod.</td>
<td>The Certification Basis for the original product remains applicable to the STC/Approval. The requirements for environmental protection and the associated certification remain unchanged and remain applicable to this certificate/Approval.</td>
<td>Master Data List, PP-022-AME-00, Issue 2, dated 23/07/2012 Aircraft Flight Manual Supplement PP-022-AMF-02, Issue 2, dated 23/07/2012 or later revisions of the above listed documents approved by EASA</td>
<td>Limited to serial numbers BL 49 and BB 14743 it is forbidden to use the modified pod to carry baggage or cargo. This modification does not include power to the antennas installed on the pod. The operation of the antennas is not included in this approval and must be approved by a separate change.</td>
<td>23/07/2012</td>
<td>Active</td>
<td>10040714</td>
</tr>
<tr>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td>STC 0038601</td>
<td>A319-115</td>
<td>VIP Cabin Interior Completion. See Master Data List BCJ03856-001.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the STC/Approval.</td>
<td>Master Data List, PP-022-AME-00, Issue 2, dated 23/07/2012 Aircraft Flight Manual Supplement PP-022-AMF-02, Issue 2, dated 23/07/2012 or later revisions of the above listed documents approved by EASA.</td>
<td>Approval valid only for MSN 3856. The aircraft is certified for private use only, not to be operated for commercial air transport. The aircraft is a No Smoking aircraft. For other limitations, refer to Airplane Flight Manual Supplement BCJ03856-201-01 Rev. C.</td>
<td>23/07/2012</td>
<td>Active</td>
<td>10040716</td>
</tr>
<tr>
<td>BOMBARDIER INC.</td>
<td></td>
<td>STC 0038621</td>
<td>BD-700-1A10</td>
<td>Installation, Aircraft Completion Modifications - ASN 9443.</td>
<td>The Certification Basis for the original product remains applicable to the STC/Approval. The requirements for environmental protection and the associated certification remain unchanged and remain applicable to this certificate/Approval.</td>
<td>Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012 - Bombardier Certification Definition Record RAE-8700-001 Revision Tabled 10 July 2012 - Bombardier Airplane Flight Manual Supplement G-FG050401 Revision B; Instructions for Continued Airworthiness document GC0443-0000-10A or later revisions...</td>
<td>Restricted to ASN 9443. The Wireless Local Area Network (WLAN) is intended to provide internetconnectivity and small services to the airplane's cabin using portable/switchable devices (PEDs). Any other intended function of this equipment requires a re-examination of the certification basis.</td>
<td>24/07/2012</td>
<td>Active</td>
<td>10040730</td>
</tr>
<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA Cert No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10040722</td>
<td>BOMBARDIER INC.</td>
<td></td>
<td>TC-20 ST-NO. SA12-31</td>
<td>BD-700-1A10-1A11</td>
<td>Modification, Electronic Wiring Monitor and Brake Temperature Monitor Unit (HBMU)</td>
<td>The Certification Basis for the original product remains applicable to this certificate of approval. The requirements for environmental protection and the associated certification noise and emissions levels of the original product are unchanged and remain applicable to this certificate of approval.</td>
<td>Transport Canada STC No. SA12-31 Issue 2 dated 25 May 2013, ACS-AV Ltd. Modification Data Summary H19000 Revision C dated 30 March 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2012/02/CF (or subsequent revisions or amendments).</td>
<td>24/02/2012</td>
<td>Active</td>
<td>10040722</td>
</tr>
<tr>
<td>10040724</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td></td>
<td>FAA STC NO. ST02744CH-D</td>
<td>BD-700-1A10</td>
<td>Installation of a Rockwell Collins Wireless Local Area Network (WLAN).</td>
<td>The Certification Basis for the original product remains applicable to this certificate of approval. The requirements for environmental protection and the associated certification noise and emissions levels of the original product are unchanged and remain applicable to this certificate of approval.</td>
<td>FAA STC No. ST02744CH-D dated 30 November 2011, Midcoast Aviation Index Set 35-490007, Revision NC dated 05 August 2009, Midcoast Aviation Airplane Flight Manual Supplement 35-490004, Revision NC dated 06 August 2009, or later revisions of the above listed documents approved by EASA in accordance with EASA ED D.</td>
<td>24/07/2012</td>
<td>Active</td>
<td>10040724</td>
</tr>
<tr>
<td>10040738</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td></td>
<td>FAA STC NO. SA1633WI</td>
<td>EMB 500</td>
<td>Installation of Concorde RG-350E VRSLA Batteries in the Embraer 500 in accordance with Concorde Battery Corporation Master Drawing List, DWG# 5-0365 Revision NC dated August 25, 2010 or later FAA approved revisions.</td>
<td>The Certification Basis for the original product remains applicable to this certificate of approval. The requirements for environmental protection and the associated certification noise and emissions levels of the original product are unchanged and remain applicable to this certificate of approval.</td>
<td>Concorde Battery Corporation Master Drawing List, DWG No. 5-0365, Revision NC, dated August 25, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2012/02/CF (or subsequent revisions or amendments).</td>
<td>None.</td>
<td>24/02/2012</td>
<td>Active</td>
</tr>
<tr>
<td>10040777</td>
<td>AIRBUS S.A.S.</td>
<td></td>
<td>A330-214</td>
<td>Airbus Corporate Jet Centre Modification CJ1108 Issue 1 - Installation of Business Class Seating Area - &quot;Gulf Air - Type B&quot;</td>
<td>In accordance with the EASA TCDS The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: - CRI A 2301-207 WJC Centre Conversion Project Certification Basis - The requirements for environmental protection.</td>
<td>Modifications Approval Sheet, ref.: CJ1108, Issue 1, dated 25th July 2012 and Airplane Flight Manual AFM Supplement, ref.: FM-5 MOD CJ1108, dated 1st July 2012 for the above listed documents approved by EASA</td>
<td>This STC is limited to MSN 5171, MSN 5175</td>
<td>26/07/2012</td>
<td>Active</td>
<td>10040777</td>
</tr>
<tr>
<td>10040786</td>
<td>JET AVIATION AG, Bazel</td>
<td></td>
<td>B737-800</td>
<td>Forward VIP Lavatory reconfiguration - New shower system - Reconfiguration of existing systems</td>
<td>JAR 25, Amdt: Change 13 except as noted in EASA TCDS IM.A.120 The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CRI D-01 Shower Installation/CM H-01 Installation Instructions for Continued Airworthiness</td>
<td>Instructions for continued airworthiness ref. B3AUH003-150 Master Data List, B3AUH003-019 Flight attendant handbook ref. VH-04 or later revisions of the above listed documents approved by EASA.</td>
<td>S/N: 33473 (A6-ALU)</td>
<td>26/07/2012</td>
<td>Active</td>
<td>10040786</td>
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<tr>
<td>10040781</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td></td>
<td>FAA STC ST01643WI</td>
<td>DHC-8-301, -311, -314, -315 DHC-8-201, -202 DHC-8-102, -103, -106</td>
<td>Installation of Concorde RG-380E/44 (42AH) or RG-380E/60L (48AH) Series valve Regulated Sealed Lead-Acid (VRSLA) batteries as replacement of OEM batteries</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation/Operation must be in accordance with Concorde Battery Corporation Master Data List (MDL) Drawing N° 5-5284-1, 5-5284-3Rev A, Operation/Installation Manual Supplement (AF MS) Drawing N° 5-5285, Rev. 3, NC Maintenance Manual</td>
<td>none</td>
<td>26/07/2012</td>
<td>Active</td>
</tr>
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</table>
1. Camera Modification Kit System

Installation of Due Talbotcom Strakes. This STC revises and supersedes, BA STC RC 1163 and the Bell Helicopter Textron Inc. Model 204B, 412 and 412EP. The AB 212 is removed as its not included in the FAA STC. Rev. 1 updates the STC holder address.

<table>
<thead>
<tr>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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</thead>
</table>

2. 204B, 205A-1

Order to extend the bins inboard). Bins located in the passenger cabin in order to extend bins inboard.

<table>
<thead>
<tr>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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</thead>
<tbody>
<tr>
<td>Day/Night VFR only, flight into known icing conditions prohibited. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of...</td>
<td></td>
</tr>
</tbody>
</table>

3. 212, 412, 412EP

FAR 23 Amdt 1-7 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.

4. 340A

The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

<table>
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<tr>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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</table>

5. B757-200

The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

<table>
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<tr>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<tbody>
<tr>
<td>Restricted to SIN 07Aa per AFM Supplement N° SAFM-FA120 Rev. 19 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness...</td>
<td></td>
</tr>
</tbody>
</table>

6. EC135 P1, EC135 P2, EC135 T1, EC135 T2, EC135 T2+

The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

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<tr>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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</table>

7. 1.0.0.0 models; RFMS-BH205-2a revision IR dated 12 March 2004 - Flight Manual Supplement for Model 205A-1 Performance Improvement; RFMS-BH205-3a revision IR dated 12 March 2004 - Flight Manual Supplement for Model 204B Performance... |


10. Avionics/Prototype only

The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

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<th>Certification basis</th>
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>10040859</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td></td>
<td></td>
<td>A300F4-600ST</td>
<td>Installation of an Inmarsat Satcom for voice and ACARS on A300-600ST</td>
<td>This Certification Basis allows for the installation of a satellite communication system for voice and ACARS on A300-600ST. The requirements for installation include ensuring electromagnetic compatibility with existing avionic systems and maintaining the aircraft's airworthiness.</td>
<td>None</td>
<td>31/07/2012</td>
<td>Active</td>
<td>10040859</td>
<td></td>
</tr>
<tr>
<td>10040868</td>
<td>SABENA TECHNICS BOO</td>
<td></td>
<td></td>
<td>A320-214</td>
<td>Installation of SATCOM system</td>
<td>This Certification Basis (CB) allows for the installation of a SATCOM system on A320-214 aircraft. The requirements for installation include ensuring proper integration with existing avionic systems.</td>
<td>- Certification Plan PC- D0088-002 Issue 4 dated 20/07/2012 - Master Drawing List MDL-S-50-00261, Issue B, dated 26/07/2012 - Instructions for Continued Airworthiness (ICA) ICA- D0088-00-001 Issue A dated 26/08/2012 - Modification Approval Sheet MA6.50-00-2261 Issue 3 dated 20/07/2012</td>
<td>31/07/2012</td>
<td>Active</td>
<td>10040868</td>
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<tr>
<td>10040889</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td></td>
<td>CESSNA 560XL</td>
<td>Installation of a Moving Map System.</td>
<td>The Certification Basis allows for the installation of a moving map system on CESSNA 560XL aircraft. The requirements for installation include ensuring proper integration with the aircraft's avionic systems.</td>
<td>Jet Aviation Master Data List: ref 2012-4-16, rev IR, dated 30 July 2012 - Jet Aviation AFM Supplement: ref 2012-4-9, rev A, dated 21 July 2012</td>
<td>31/07/2012</td>
<td>Active</td>
<td>10040889</td>
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<tr>
<td>10040870</td>
<td>ORCON CORPORATION</td>
<td></td>
<td>B737-200/-200C/-300/-400</td>
<td>-500</td>
<td>Thermal/aoustic insulation system for Boeing 737 series aircraft insulation blankets installation.</td>
<td>The Certification Basis allows for the installation of a thermal acoustic insulation system for Boeing 737 series aircraft. The requirements for installation include ensuring proper integration with existing avionic systems.</td>
<td>Master Drawings List MDL-4000 Revision H(+)= &gt;Continued Airworthiness Instructions CAI-4000, Revision E(+)&gt;</td>
<td>31/07/2012</td>
<td>Active</td>
<td>10040870</td>
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<tr>
<td>10040891</td>
<td>ASI INNOVATION</td>
<td></td>
<td>P2006T</td>
<td>ASI-0084-03 dated 30 July 2012 - Installation of removable misson consoles (without equipment) in the aircraft cabin incapacitation passenger seat.</td>
<td>The Certification Basis allows for the installation of removable mission consoles (without equipment) in the aircraft cabin incapacitation passenger seat. The requirements for installation include ensuring proper integration with existing avionic systems.</td>
<td>None</td>
<td>20/09/2013</td>
<td>Active</td>
<td>10040891</td>
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<tr>
<td>EASA STC No.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10040901</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>2</td>
<td>FAA STC ST02907CH</td>
<td>B747-400F</td>
<td>Installation of ECS EZMount mechanical mounting assemblies in accordance with Electronic Cable Specialists, Inc. Master Data List ECS 290135; Revision H, dated November 29, 2011, or later approved revision/after of FAA STC ST02907CH</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. 14 CFR Part 25 Amendment 25-119 for the following requirements: 25.301(a)(6); 32.103; 38.305(a)(b); 38.307(A); 25.562(c); 25.625(b); 25.665(b); 25.662(b); 25.1245; 38.307(A).</td>
<td>Master Data List ECS 290135, Revision H, dated November 29, 2011 or later approved revision/approval of FAA STC ST02907CH</td>
<td>02/08/2012</td>
<td>Active</td>
<td>10040901</td>
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<tr>
<td>10040905</td>
<td>NEXTANT AEROSPACE L.L.C.</td>
<td>400A</td>
<td>FAA STC ST03960AT</td>
<td>400A</td>
<td>Installation of Rockwell Collins Vue-Flight Entertainment (FVE) Aircraft Access Cabin II Iridium Phone and High Speed Internet System in the Wireless Local Area Network (WLAN) Astronomics 115V AC 80 Hz Cabin Inverter, and Emteq LED Cabin Lighting System.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. FAR 25 Amendment as defined in FAA STC ST03960AT. The requirements for environmental protection and the associated certificated noise...</td>
<td>Installation in accordance with Nextant Aerospace LLC Master Drawing, dated No. DB-100267 Revision E, dated January 26, 2012; Operation in accordance with Nextant Aerospace LLC FAA Approved Flight Manual Supplements. When Rockwell Collins Vue System is installed Airplane Flight Manual/Supplement DBI...</td>
<td>None.</td>
<td>02/08/2012</td>
<td>Active</td>
<td>10040905</td>
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<tr>
<td>10040906</td>
<td>NEXTANT AEROSPACE L.L.C.</td>
<td>400A</td>
<td>FAA STC ST03110CH</td>
<td>400A</td>
<td>Installation of L3 Communications Digital Flight Data Recorder</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. FAR 25 Amendment as detailed on FAA STC ref ST03110CH. The requirements for environmental protection and the associated certificated noise...</td>
<td>Installation in accordance with Nextant Aerospace LLC Master Drawing, dated No. NX1111001 Revision 1, dated May 29, 2012; Operation in accordance with Nextant Aerospace LLC FAA Approved Flight Manual Supplement No. NX1111004 Revision IR, dated May 17, 2012 or later revisions of the above listed documents</td>
<td>None.</td>
<td>02/08/2012</td>
<td>Active</td>
<td>10040906</td>
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<td>10040912</td>
<td>DECA AVIATION ENGINEERING LTD, aka PAL AEROSPACE</td>
<td>REV. 1</td>
<td>TCCA STC NO. O-LSA00-8295/D</td>
<td>BD-700-1A10</td>
<td>Baggage Door Locking System Modification</td>
<td>EASA E.M A.309. The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC No. O-LSA00-8295/D Issue 2 dated 04 January 2002 DECA Modification Summary MS00093938 Issue 1 dated 09 November 2000 DECA Maintenance Manual Supplement DECA-4885-02 Revision 2, TC accepted 19 December 2001 or later revisions of the above listed documents approved by EASA in accordance...</td>
<td>Restricted to ASN 9003</td>
<td>28/08/2012</td>
<td>Active</td>
<td>10040912</td>
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<td>10040913</td>
<td>SNQ CERTIFICATION SERVICES Ltd.</td>
<td>REV. 5</td>
<td>CAA UK RESPONSIBILITY</td>
<td>747-8</td>
<td>Installation of the Aerolift™ Overhead Sleeping Compartment immediately forward of the Overhead Crew Rest. Two following configurations are certified: (1287, 1036, 1044, 1532), (1036, 1287 and 1532) (issue 3.1. For 1287 and 1532, the following peculiarities are included: IFE and PC Po...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (a): D-100 “Installation of overhead sleeping compartment” issue 2.61-7(1) “EASA certification basis definition” issue 27.2. The requirements for environmental...</td>
<td>Master Data List ECS 209513, Revision H, dated November 29, 2011 or later approved revision/approval of FAA STC ST02907CH</td>
<td>26/10/2020</td>
<td>Active</td>
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<td>STC Holders</td>
<td>Rev.</td>
<td>Certification basis</td>
<td>Description</td>
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<tr>
<td>IMNOTECH AVIATION DIVISION OF IMP GROUP LTD</td>
<td>REV. 1</td>
<td>CTOA-STD-C-LSA12-105</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9433</td>
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<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U.</td>
<td>REV. 2</td>
<td>B200GT</td>
<td>&quot;Stretcher kit Aviaf 32-0117&quot;: Installation of a stretcher kit replacing LH seats (rows 2 and 3), according to Certification Plan &quot;NAER.M1204425-PNC&quot;, Revision 0, Date: 27/07/2012. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>HAWKER BEECHCRAFT Ltd</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>C900GT</td>
<td>To introduce L3 Electronic Standby Instrument System (ESIS) in lieu of existing Meggitt Mk II Secondary Flight Display (SFD) System (the existing Meggitt Mk II SFD (part number 39221-01-01) is removed and replaced with L3 GH.000.2 ESIS (Partnumber 39200-0400-0401). The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>REV. 1</td>
<td>737-800, 700, 800, 900, 900ER</td>
<td>Cabin Altitude and Take-off Configuration warnings and indications - Activation. The Certification Basis for the original product remains applicable to this certificate/approval. The applicant elected to comply with CS 25 Amendment 11. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC</td>
<td>ST00600CH-D</td>
<td>As per FAA STC ST00600CH-D issued dated 12 Sept 2008, for the installation of Honeywell Enhanced Ground Proximity Warning System with WindShear Detection and optional Terrain Display (55W level - 216-218) and optional Runway Advisory Awareness System (RAAS). The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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**List of EASA Supplemental Type Certificates**

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<tr>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>F&amp;M ENTERPRISES INC.</td>
<td>FAA STD SE75695W</td>
<td>2. C090 SERIES, O-200 SERIES</td>
<td>1. O-75 SERIES, C-65 SERIES</td>
<td>Install adapter to permit use of a spin-on oil filter in accordance w/ F. &amp; M. Enterprises' Installation Instructions for Model TAF-Long Oil Filter, dated October 26, 1998, or later FAA approved revision EASA Validation of FAA STC SE75695W</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAAD system to FAA approved drawing package MA2174, revision IR dated 06-Sep-2002 and Installation Instructions dated 27-Aug-2003 or later FAA approved revision.</td>
<td>Engines modified in accordance with this STC will not differ in performance or operating limitations for unmodified engines.</td>
<td>06/08/2012</td>
<td>Active</td>
<td>10040937</td>
</tr>
<tr>
<td>F&amp;M ENTERPRISES INC.</td>
<td>FAA STC SE90356SC</td>
<td>3. C5-520 SERIES</td>
<td>2. C-475 SERIES</td>
<td>Install F. &amp; M. Enterprises' Model C5-LC, C5-LC-5, C5-LC-5K oil filter adapter for a spin-on oil filter in accordance with F. &amp; M. Enterprises Installation Instructions dated October 26, 1998, or later FAA approved revision. EASA Validation of FAA STC SE90356SC</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Installation Instructions for Model CO-330 dated October 26 (1999), or later FAA approved revision.</td>
<td>None</td>
<td>06/08/2012</td>
<td>Active</td>
<td>10040938</td>
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<tr>
<td>F&amp;M ENTERPRISES INC.</td>
<td>FAA STC SE84095W</td>
<td>5. T90-360 SERIES</td>
<td>3. G-330 SERIES</td>
<td>Installation of F. &amp; M. Enterprises Model CO-330 or C6SC spin-on oil filter adapter, in accordance with F. &amp; M. Enterprises Installation Instructions for Model CO-330 dated October 26, 1998, or later FAA approved revision. EASA Validation of FAA STC SE84095W</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Installation Instructions for Model CO-330 dated October 26 (1999), or later FAA approved revision.</td>
<td>Engines modified in accordance with this STC will not differ in performance or operating limitations for unmodified engines.</td>
<td>06/08/2012</td>
<td>Active</td>
<td>10040939</td>
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<tr>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SA01161SE</td>
<td>REV. 2</td>
<td>PA-325-300, PA-32R-300</td>
<td>Installation of Vortex Generators on the wing and tail surface according to drawing package MA2174, revision IR dated 06-Sep-2002 and Installation Manual MA2175 revision A dated 27-Aug-2003, or later FAA approved revision to either of these documents.</td>
<td>The Certification Basis of the original product remains applicable to this certificate/approval.</td>
<td>FAA approved drawing package MA2174, revision IR dated 06-Sep-2002. Installation Manual MA2175 revision A dated 27-Aug-2003.</td>
<td>If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. If more than 5 vortex generators are damaged or missing, the A/C is not airworthy.</td>
<td>11/09/2012</td>
<td>Active</td>
<td>10040940</td>
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<tr>
<td>LEARJET Inc.</td>
<td>FAA STC ST011225SE-D</td>
<td>REV. 1</td>
<td>LEARJET 35/31A/36/36A</td>
<td>Group Aircraft Approval for RVSM Validation of FAA STC ST011225SE-D dated 14 April 2010. Note: This approval constitutes an update to EASA STC ref. N° 1426 dated December 2003 (LBASTC TA0751, dated 27 November 2003) in order to broaden the compatibility of this STC's installation with other STCs/aircraft...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Approval for Continued Airworthiness, Rev. B, dated May 19, 2004 - Airplane Flight Manual Supplement Aircraft Aeromech document AW-03-ALJ3536Rev &quot;original&quot;, FAA approved May 9, 2003 - I...</td>
<td>The following items listed in the Limitations and Conditions section of the FAA STC ST011225SE-D dated April 14 2010, are excluded from EASA approval of the STC, because they are related to a configuration of &quot;C-14A&quot; that has not been validated by EASA § 4 Group of STCs according to table, the &quot;ST00033MC...</td>
<td>07/08/2012</td>
<td>Active</td>
<td>10040951</td>
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**Legend:**
- STC: Supplemental Type Certificate
- FAA: Federal Aviation Administration
- EASA: European Union Aviation Safety Agency
- STCs: Supplemental Type Certificates
- ST011225SE-D: Document number
- ST00033MC: STC number
- FAA STC: Federal Aviation Administration Supplemental Type Certificate
- FAA STC Approval: FAA approval for Continued Airworthiness
- FAA STC Approval: FAA approval for Continued Airworthiness, Rev. B
- FAA STC Approval: FAA approved drawing package MA2174, revision IR dated 06-Sep-2002
- FAA STC Approval: FAA approved drawing package for Model CO-330 dated October 26 (1999), or later FAA approved revision.
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<tr>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NA</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis for original product remains applicable to this certificate/approval. Requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</th>
<th>Change Document Supplemental Type Certificate Doc. No. STC-AT-050.Rev. 3</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td></td>
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<td>P.68 C, P.68 C-TC</td>
<td>Title: Mission cabin configuration P.68 C. Description: This STC consists of 3 steps: 1. To install different mission equipment in a Vulcanair P.68 aircraft, step 1 modifies the aircraft cabin and includes provisions to accommodate Mission Equipment. Step 2: Optional installation of an equipment module...</td>
<td>Implementation of provisions for AEF Class 2, NA, VIB C2 in accordance with Aero Vodochody M/DL AM003-ML Rev. 4. Rev. 1. Installation of power switches to the 28V DC bus for safe poweroff for the EFB system. Rev. 2: Name change of the DOA, Rev. 3: Transfer of STC.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. Requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Vulcain SB-163 must be carried out prior to this STC.</td>
<td>07/02/2012</td>
<td>Active</td>
<td>10040972</td>
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<tr>
<td>ANLUVI</td>
<td>REV. 3</td>
<td></td>
<td>A310-322/-324/-326/-328</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): CRI H-01 Enhanced Airworthiness Programme for Airplanes system – Instructions for Continued Airworthiness on EWAIS. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this approval.</td>
<td>Documents for initial issue of this approval: Aero Vodochody Maintenance Manual Supplement AM003-ML Rev. 03, Aero Vodochody Aircraft Flight Manual Supplement AM003-AM Rev. 05, Aero Vodochody Wiring Diagram Supplement AM003-MD Rev. 03, Aero Vodochody Instruction for Continued Airworthiness AM003-C, Rev. 0.</td>
<td>For MSN 0554, 0560, 0814 and 0667. Revision 1 of this approval is valid only under the following conditions: the completion of installation of parts according to Aero Vodochody Minor Change A-42 – STC-2 0058 for MSN 0554, A-STC-2 0059 for MSN 0560, A-STC-2 0060 for MSN 0814 and A-STC-2 0061 for MSN 0667.</td>
<td>05/05/2021</td>
<td>Active</td>
<td>10040961</td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td></td>
<td>FAA STC</td>
<td>B767-200/-300-300F/-400ER</td>
<td>Installation of BE Aerospace Inc. Vacuum Toilet Assembly Retrofit Kit.</td>
<td>The Certification Basis for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this approval.</td>
<td>BE Aerospace Installation Instructions, AMM 38-30-07, Revision 01, 11/08/2011; BE Aerospace ICA document, AMM 38-30-08, Revision 01, 11/04/2011; later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CIF (or subsequent revisions of this decision).</td>
<td>Prior to installation of this modification it must be determined that there is an interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effects on the airworthiness of the product.</td>
<td>06/04/2012</td>
<td>Active</td>
<td>10040966</td>
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<tr>
<td>TIMCO AEROSYSTEMS</td>
<td></td>
<td>FAA STC</td>
<td>777-300</td>
<td>Audio/Video On-Demand &amp; In-seat Power System Installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>TIMCO AEROSYSTEMS Instructions for Continued Airworthiness 11T133R0509, Rev. A dated September 29, 2011; later revisions of the above listed document approved by EASA in accordance with EASA ED Decision 2004/04/CIF (or subsequent revisions of this decision).</td>
<td>None.</td>
<td>10/06/2012</td>
<td>Active</td>
<td>10040972</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
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<td>ATP</td>
<td>FMS Upgrade for P-RNAV</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Scandinavian Avionics Engineering Order 2071338-ED Issue 1 dated 05/12/2012; Right Manual Supplement 2071338-AM Issue 1 dated 09/06/2012; Continued Airworthiness 2071338-CA Issue 1 dated 05/12/2012; later revisions of the above listed document approved by EASA.</td>
<td>None.</td>
<td>10/06/2012</td>
<td>Active</td>
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<td>STC Ref.</td>
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<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>JET AVIATION AG, Basel</td>
<td>10040974</td>
<td>REV. 1</td>
<td>AS330-22D, AS330-243, AS330-201, AS330-242, AS330-203</td>
<td>EMS Satcom AMT-3800 antenna installation</td>
<td>This modification installs EMS Satcom AMT-3800 high gain antenna, P/N 4393, on the aircraft. It also includes installation of an associated Low Noise Amplifier, and associated structural modifications.</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and associated(certified) noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Master Data List ref. BA33023261-001, Revision 00, for Continued Airworthiness, ref. BA33023261-150-01, Revision 00, on Weight &amp; Balance Statement, ref. ASI-SDL-0087-02, dated 01/08/2012.</td>
<td>Prior to installation of this design change it must be determined that there is no introduction of any adverse effects upon the airworthiness of the product. This Supplemental Type Certificate is approved on none.</td>
<td>23/01/2013</td>
<td>Active</td>
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<tr>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>10040981</td>
<td>REV. 1</td>
<td>A340-541</td>
<td>Oven installation</td>
<td>Description as per:ASI</td>
<td>ASI INNOVATION</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated(certified) noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>JAR 25, Amtd: 14 The Certification Basis for the original product and the following: A-1301-003, Existing Exemption &amp; Deviation, ES/DF-121-003, Equipment Substitutions...</td>
<td>01/08/2012</td>
<td>Active</td>
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<tr>
<td>ASI INNOVATION</td>
<td>10040999</td>
<td>REV. 1</td>
<td>EMB-135LR</td>
<td>Oven installation</td>
<td>Description as per:ASI SDL, STC Data List ref. ASI-SDL-0087-03, dated 01/08/2012; ASI-MDL, Master Drawing List ref. ASI-MDL-0087-01, dated 01/08/2012; ASI-TechnicalDossier ref. ASI-TD-0087-02, dated 01/08/2012.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated(certified) noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>ASI SDL, STC Data List ref. ASI-SDL-0087-03, dated 01/08/2012; ASI-MDL, Master Drawing List ref. ASI-MDL-0087-01, dated 01/08/2012; ASI-TechnicalDossier ref. ASI-TD-0087-02, dated 01/08/2012 or later revisions of the above listed documents approved by EASA.</td>
<td>Applicability limited to EMB-135LR S/N 886</td>
<td>14/02/2012</td>
<td>Active</td>
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<tr>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>10041017</td>
<td>REV. 1</td>
<td>FAA STC SR02569Y</td>
<td>Installation of Sky Connect Tracking System in accordance with MasterDrawing List MDL07094 Revision 1 of the STC is issued to reflect a change of STC Holder.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated(certified) noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Definition and installation, Inspection and Maintenance-Avionics Design Services Ltd. MasterDrawing List MDL07094 Rev. C, dated July 02, 2008. Airworthiness, ref. ASI-SDL-0087-03, dated 01/08/2012 or later revisions of the above listed documents approved by EASA.</td>
<td>This STC is not applicable to the AS350B3 model helicopters with modification OP4305 (Aerial 2D engine installation) incorporated. Prior to installation of this design change it must be determined that there is no introduction of any adverse effects upon the airworthiness of the product.</td>
<td>29/10/2015</td>
<td>Active</td>
<td>10041017</td>
</tr>
<tr>
<td>HOWARD J. FULLER, JR.</td>
<td>10041028</td>
<td>REV. 1</td>
<td>FAA STC SR00868B</td>
<td>R44</td>
<td>Operation of rotorcraft on unleaded automotive gasoline.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated(certified) noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Howard J. Fuller, Jr., Rotorcraft Flight Manual Supplement No. 441 Rev. 2, dated July 25, 2002 or later revisions of the above listed documents approved by EASA.</td>
<td>1. This approval is limited to those model helicopters equipped with pylon O-540- P185 engine modified AIV STC 8003599W (EASA.IM.E.S. 0102) that is applicable only with the following engine power ratings,– Maximum Continuous Power 255 HP (24.7 in. Hg.) @ 2600 RPM, Takeoff 255 HP (28.3...</td>
<td>17/08/2012</td>
<td>Active</td>
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<tr>
<td>BOMBAIRDIER INC.</td>
<td>10041031</td>
<td>REV. 1</td>
<td>TCCA STC NO. B-1902-400</td>
<td>Chairman Interior Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated(certified) noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC NO. C-TCA1-450 Issue 6 dated 29 June 2012 DECA Aviation Engineering Limited Modification Summary MS0100, Revision 1.</td>
<td>17/06/2012</td>
<td>Active</td>
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<td>STC Holder</td>
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<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>AERODISA S.L.</td>
<td>10041036</td>
<td></td>
<td></td>
<td>B757-200</td>
<td>Cabin Re-configuration and Airvod IFE System Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate. Additional requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this approval.</td>
<td></td>
<td>None</td>
<td>17/08/2012</td>
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<tr>
<td>AERODISA S.L.</td>
<td>10041044</td>
<td></td>
<td></td>
<td>737-800</td>
<td>VIP Cabin Interior Completion B82</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. Additional requirements are detailed in CRI A-01 Issue 05, in particular: Special Conditions: CRI D-02 Isolated compartments CRI D-10 (3)</td>
<td></td>
<td>17/08/2012</td>
<td>Active</td>
<td>10041044</td>
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<tr>
<td>AVIONIK ENTWICKLINGS GmbH</td>
<td>10041049</td>
<td></td>
<td></td>
<td>DA40, DA40D, DA40F</td>
<td>Dual/Single Installation of Garmin G/PS COMMNAV GNS430W / GNS530W.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. CS-23, Amendment 2. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement ASR-2012-052: FMS-05-00, Rev. 00. or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>17/08/2012</td>
<td>Active</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10041062</td>
<td></td>
<td></td>
<td>GV-SP</td>
<td>Installation of an Airplane Interior</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. CRI D-60 &quot;Flight deck deformation into the overwing Emergency Exit and Emergency Egress.&quot; - Equivalent Level of Safety/Min-CRI F-55 &quot;in cabin...</td>
<td>- FAA STC ST03771AT-D issued 8th May 2010. Installation in accordance with Gulfstream Aerospace Corporation Index List GCS02005266, Revision C, dated May 5, 2010. - Operation to be in accordance with Gulfstream Flight Manual Supplement GCS02005354 and GCS02005353, Issue IR, dated May 5, 2010.</td>
<td>Limitations are defined in document GC520040593, &quot;Follow-On Installation Guidance&quot;, Revision NC, dated May 5, 2010. The installer must determine whether this design change is compatible with previously approved modifications. If the holder agrees to permit another person to use this certificate to alter an aircraft, the holder must agree to permit another person to use this certificate to alter an aircraft.</td>
<td>20/08/2012</td>
<td>Active</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10041063</td>
<td></td>
<td></td>
<td>GV</td>
<td>Installation of a Transponder Flight Identification</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- FAA STC ST02864AT-D, amended 2nd October 2009-Installation in accordance with Gulfstream Aerospace Corporation Index List CES1 1289000, Revision J, dated October 21, 2009. Aircraft Flight Manual Supplement CES1 1289000, Revision NC., dated July 12, 2004.</td>
<td>None</td>
<td>20/06/2012</td>
<td>Active</td>
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<td>AERODISA S.L.</td>
<td>10041087</td>
<td></td>
<td></td>
<td>DC-9-87, MD-88, DC-81, DC-82, DC-9-83</td>
<td>Equipment update for MD80 series SalCom System - Update for MD80 SeriesSatCom System approved under EASA STC</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. CRI H-00. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Aerospace Doc., AE020111_14_MDI, Master Data Link AE020111_14, AFMS Aircraft Flight Manual Supplement System 148M01, Rev...</td>
<td>This STC is applicable only to aeroplanes that have already incorporated Aerodisa STC certified as EASA STC 10031538 dated 24 August 2010.</td>
<td>20/06/2012</td>
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<td>STC Holder</td>
<td>Description</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>Installation of High Frequency (HF) Antenna Kit (PN 429-007-001) for Bell Helicopter Textron Model 429 Helicopters.</td>
<td>- Copy of the Installation Manual, dated June 30, 2011.</td>
<td>- The installation of the HF Antenna Kit is included in this STC.</td>
<td>None.</td>
<td>22/08/2012</td>
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<td>FAA STC SR09677REC-D</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>Installation of Sliding Door Push-Out Window Kit (PN 429-563-001/2/3/4/5/6/7/8/9) for Bell Helicopter Textron Model 429 Helicopters.</td>
<td>- Copy of the Installation Manual, dated August 14, 2011.</td>
<td>- The installation of the Sliding Door Push-Out Window Kit is included in this STC.</td>
<td>None.</td>
<td>22/08/2012</td>
<td>Active</td>
<td>10041097</td>
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<td>FAA STC SR09707DE</td>
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<td>SNAKY CREEK ENTERPRISES, LLC/a</td>
<td>F.A.A Approved revisions.</td>
<td>FAA STC SA0246MCH</td>
<td>- Approval of the FAA STC SA0246MCH dated August 14, 2011.</td>
<td>None.</td>
<td>03/07/2012</td>
<td>Active</td>
<td>10041098</td>
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<td>SPECTRUM AEROMED</td>
<td>Air Ambulance Conversion Install an ambulance conversion in accordance with Spectrum Aeronautics, Master Drawing List 60450, dated July 23, 2007, or later FAA approved revisions.</td>
<td>- Copy of the Installation Manual, dated August 14, 2011.</td>
<td>- The installation of the Air Ambulance Conversion is included in this STC.</td>
<td>Likely to be in accordance with FAA STC SA0246MCH dated August 14, 2011.</td>
<td>22/08/2012</td>
<td>Active</td>
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<td>FAA STC SA0246MCH</td>
<td>PC-12/47 PC-12/45 PC-12 PC-12</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>Executive Completion, Multiple STC.</td>
<td>FAA STC ST0426AT-D</td>
<td>- Copy of the Installation Manual, dated August 14, 2011.</td>
<td>Limitations are defined in document GC52204608AT-D.</td>
<td>21/08/2012</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
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<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>BELL HELICOPTER TECHTRON, Inc.</td>
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<td>FAA SR00741DE</td>
<td>BELL 429</td>
<td>Removable Crew Door Installation Kit (FY N 429-650-001/34) for Bell Helicopter</td>
<td>The Certification Basis for the original product remains applicable to this certificate/</td>
<td>Flight Manual Supplement, Document No. AA-1070, Revision NR, dated January 19, 2012.</td>
<td>None</td>
<td>22/02/2012</td>
<td>Active</td>
<td>10041125</td>
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<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>204B</td>
<td>Exchange of engine torque and temperature indicator.</td>
<td></td>
<td>The Certification Basis for the original product and the following additional alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>127427-01-069 revision 0 dated 13 August 2012; Flight Manual Supplement/Applicable revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/ACF (or subsequent revisions of this decision).</td>
<td>none</td>
<td>23/08/2012</td>
<td>Active</td>
<td>10041143</td>
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<tr>
<td>KAMAN CORPORATION/RCRAFT WHEEL AND BRAKE, LLC</td>
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<td>FAA STC SH826NE</td>
<td>AS35EF1</td>
<td>Installation of Parker 1C44-2 Fuel Pump</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Instructions for 1C44-2 Fuel Boost Pump-revision B dated 17 August 2003.</td>
<td>None</td>
<td>23/08/2012</td>
<td>Active</td>
<td>10041145</td>
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<tr>
<td>KAMAN CORPORATION/RCRAFT WHEEL AND BRAKE, LLC</td>
<td></td>
<td>FAA STC SR153CH</td>
<td>AS 332 L, AS 332 L1</td>
<td>Installation of Parker 1064-15 Fuel Pump</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Instructions for 1064-15 Centrifugal Fuel Boost Pump -revision dated 25 March 2002.</td>
<td>None</td>
<td>24/08/2012</td>
<td>Active</td>
<td>10041148</td>
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<tr>
<td>MILLENIUM CONCEPTS, INC. REV. 1</td>
<td>FAA STC SA01671WI - D</td>
<td>2- G58 - BARON 1- G58 - BOBANZA</td>
<td>Installation of interior configuration per Master Drawing List 2033-MDL040, rev. P, dated 8 February 2012.</td>
<td>Design non-significant change. The Certification Basis of the original product and the following additional airworthiness requirements are applicable to this change. As defined in the STC certification basis approved by FAA. Environmental Protection: The certificated noise and/or emissions levels of the...</td>
<td>FAA approved Master Drawing List 2033-MDL040, rev. P, dated 8 February 2012; FAA Approved AFM Supplement 2033-AMFS-S1 dated 8 February 2012 (For G38BBobanza); FAA Approved AFM Supplement 2033-AMFS-S2 dated 8 February 2012 (For G38BBobanza)</td>
<td>FAA approved Master Drawing List 1606-MDL040, rev. P, dated 8 February 2012; FAA Approved AFM Supplement 2033-AMFS-S2 dated 8 February 2012 (For G38BBobanza); INSTALLATION MANUAL 7, dated 8 February 2012 (For G38BBobanza)</td>
<td>If the holder agrees to permit another person to use this certificate/approval the holder shall give the other person written evidence of that permission.</td>
<td>27/08/2012</td>
<td>Active</td>
<td>10041159</td>
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<tr>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td></td>
<td>B96, B96A, B96A, E56, 96-C55A, D55, D55A, E56, E56A, 96-B55A, 96-B55B, 96-C55B, 96-A56, 96-A56, 56TCL, 96-TCL, 96-A, 58A, 96B,</td>
<td>Installation of Garmin GNS 430 W system, GA 35 GPS, antennas/CMMA-341 341adupanel and Mtd Continent MD 41 annunciator according to SEAM Modification n°3823/12.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. The requirements for Environmental Protection: The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>SEAM Flight Manual Supplement n°SEAM-SMV 3823/12, revision 0 dated 02/05/2012; SEAM Modification n°3823/12 revision dated 02/05/2012 (or subsequent revisions of the above listed documents approved by EASA).</td>
<td>Installation of this design change must be performed by authorised SEAM Technical personnel. This installation is approved for BRNAV operations without coupling to an autopilot. Use of GPS for SID/STAR and approach procedures is not authorised. So software upgrades introducing new functions at equipment level...</td>
<td>24/08/2012</td>
<td>Active</td>
<td>10041160</td>
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List of EASA Supplemental Type Certificates

| STC Holder | NATIVE AVIONICS UPGRADE FOR A109A | Wingtip pyrotechnics & LED lights. | AS per: Flight manual supplement AFMS-1306 & Instructions for Continued Airworthiness ICA-1306. | 24/08/2012 | Active | 10041151 |
| STC Holder | CAA UK RESPONSIBILITY | P2006T | Installation of Avidyne Corporation | 10041170 | BOURNEMOUTH AVIATION (CONSULTANTS) Ltd. | REV. 1 | A109A II | AVIONICS UPGRADE FOR A109A | Wingtip pyrotechnics & LED lights. | Certification Basis for the original product remains applicable if threshold certificate/approval CS - 23as per TCDS EASA.A.165 Major. | Certification Basis for the original product remains applicable if threshold certificate/approval. The requirements for environmental protection and the associated certification remain. As indicated in FAA STC SA00282BO Enhanced Vision System (EVS) is not included in the scope of the approval/compatibility with Mode S alerts must be shown when a TAWS Class A is installed. | ONE PLACARD WITH THE WORDS "OPS LIMITED TO VFR ONLY" POSITIONED ON THE INSTRUMENT PANEL NEAR THE KMD 150. | 03/09/2012 | Active | 10041171 |
| STC Holder | DASSAULT FALCON SERVICE | MYSTERE-FALCON 50 | Mode S transponder upgrade to comply with Elementary and Enhanced Surveillance. Honeywell transponder MST-678A is upgraded and ATC/TCAS Control Box is replaced by new Honeywell PS-578A unit. | 28/08/2012 | Active | 10041182 |
| STC Holder | AVIDYNE CORPORATION | FAA STC SA00282BO | Installation of Avidyne Corporation 700-0008-XXX Entegra Integrated Flight Display System on Cirrus SR20 and SR22 according Avidyne Doc. NoAVFD-006 in latest FAA approved Revision | 28/08/2012 | Active | 10041185 |
| STC Holder | THALES AVIONICS Inc. | FAA STC ST0431LA | Installation of Thales Avionics IFSS TOPARTEX In-flight Entertainment(PFE) System | 29/08/2012 | Active | 10041205 |
## List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
<th>FAA STC</th>
<th>STC No.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Document</th>
<th>Limitations/Conditions</th>
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<th>System Record</th>
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<tr>
<td><strong>BELL HELICOPTER TEXTRON, Inc.</strong> 10041215</td>
<td>FAA STC</td>
<td>SR09641RC</td>
<td>407</td>
<td>Passenger Sliding Doors Kit P/N 407-270-003 for Bell Helicopter Textron Model 407 Helicopters. Reference FAA STC SR0941RC.</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated/certified noise and emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- Rotorcraft Flight Manual Supplement for Passenger Sliding Doors Kit P/N 407-270-003, Report No. AA-10101 Rev. NR, dated April 10, 2010, or subsequent revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/48/CF (or subsequent revisions of this decision). - Master Dr.</td>
<td>30/08/2012</td>
<td>Active</td>
<td>10041215</td>
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<tr>
<td><strong>BELL HELICOPTER TEXTRON, Inc.</strong> 10041217</td>
<td>FAA STC</td>
<td>SR09640RC</td>
<td>407</td>
<td>Crew Area Hardpoints Kit BHTI P/N 407-438-001/2/3 External Hardpoint Kit BHTI P/N 407-230-100/2/3 and P/N 407-230-100/10Pylon Hardpoints Kit BHTI P/N 407-638-001/2/3 Reference FAA STC SR09640RC</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated/certified noise and emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- Rotorcraft Flight Manual Supplement for External Hardpoint Kit P/N 407-230-100, Report No. AA-10114 Rev. NR, dated April 10, 2012, or subsequent revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/48/CF (or subsequent revisions of this decision). - Master Dr.</td>
<td>30/08/2012</td>
<td>Active</td>
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<td><strong>GULFSTREAM AEROSPACE CORPORATION</strong> 10041234</td>
<td>FAA STC</td>
<td>ST01105LA-D</td>
<td>GV</td>
<td>Installation of a 13 Passenger Cabin Interior</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated/certified noise and emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- FAA STC ST01105LA-D, issued 26th June 2001: Installed in accordance with Gulfstream Aerospace Corporation Index List ref. GCS2000202, Revision A, dated 25th June 2001, or subsequent revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/48/CF (or subsequent revisions of this decision).</td>
<td>31/08/2012</td>
<td>Active</td>
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<tr>
<td><strong>STANDARD AERO</strong> 10041237</td>
<td>FAA STC</td>
<td>ST00308CH-D</td>
<td>GV</td>
<td>Reconfiguration of a Cabin Foorplan Arrangement</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated/certified noise and emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- FAA STC ST00308CH-D, issued 19th December 2011: Installed in accordance with Standard Aero Master Data List - MDL122010, Revision H, dated 19th December 2011, or subsequent revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/48/CF (or subsequent revisions of this decision).</td>
<td>05/09/2012</td>
<td>Active</td>
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<tr>
<td><strong>BOMBARIDER INC.</strong> 10041238</td>
<td>TCCA STC</td>
<td>C-LSA05-104D</td>
<td>BD700-1A10</td>
<td>Cabin Interior Completion / Furnishings</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated/certified noise and emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Ref.: Limited Supplemental Type Certificate by TCCA No. C-LSA05-104/DM-S: FAA STC ST03837/CH-D, issued 15th December 2010: Installed in accordance with TCCA Master Data List - MDL120120, Revision H, dated 19th December 2011, or subsequent revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/48/CF (or subsequent revisions of this decision).</td>
<td>17/06/2012</td>
<td>Active</td>
<td>10041238</td>
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**Notes:**
1. The Passenger Sliding Doors Kit P/N 407-270-003 modified the hinged passenger doors for the Bell Helicopter Textron Canada Limited (BHTC) Model 407 helicopters. A Personal attachment fixture mounted in the passenger compartment is not intended to secure cargo or support rappelling operations 2. Inc...
3. The present STC makes provisions for...
<table>
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<tr>
<th>STC Holder</th>
<th>STC No.</th>
<th>STC Title</th>
<th>Rev. Level</th>
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<td>BOMBARDIER INC</td>
<td>10041239</td>
<td>Limited Supplemental Type Certificate</td>
<td>Rev 1</td>
<td>TCJA STC-LSA05-169D</td>
<td>Passenger Occupancy of the Crew Rest Area</td>
<td>Certification Basis for the original product remains applicable to aircraft</td>
<td>Restricted to ASN 9147</td>
<td>17/02/2012</td>
<td>Active</td>
<td>10041239</td>
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<tr>
<td>BOMBARDIER INC</td>
<td>10041261</td>
<td>Airbus Helicopters, Inc.</td>
<td>Rev 1</td>
<td>FAA STC ST01941E</td>
<td>Installation of Complete Customized Aircraft Interior</td>
<td>The Certification Basis for the original product remains applicable to aircraft</td>
<td>- For those aeroplanes that were passenger-to-freight converted in accordance with the initial issue or revision 1 of this STC -Before further flight after 13 August 2014, the three A300B4-600 series model EASA STC no. 10041267</td>
<td>13/08/2014</td>
<td>Active</td>
<td>10041261</td>
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<tr>
<td>BOMBARDIER INC</td>
<td>10041271</td>
<td>Bucher Leichtbau AG</td>
<td>Rev 1</td>
<td>FOCA STC 25.20.42 ISSUE 5</td>
<td>SEE APPROVED MODEL LIST</td>
<td>The Certification Basis for the original product remains applicable to aircraft</td>
<td>- For those STCs that were passenger-to-freight converted in accordance with the initial issue or revision 1 of this STC -Before further flight after 13 August 2014, the three A300B4-600 series model</td>
<td>15/02/2013</td>
<td>Active</td>
<td>10041271</td>
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<td>STC Holder</td>
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<td>BRITISH AIRWAYS PLO, BRITISH AIRWAYS ENGINEERING &amp; FICE OF AIRWORTHINESS</td>
<td>CAA UK RESPONSIBILITY</td>
<td>747-400</td>
<td>B747-400 TCAS change 7.1 update</td>
<td>The Certification Basis for the original product remains applicable. Issuance of certificates approval.</td>
<td>C25 Amendment 11The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>none</td>
<td>03/03/2012</td>
<td>Active</td>
<td>10041273</td>
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<tr>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CORPORATION</td>
<td>CL-600-2B16</td>
<td>CPDLC System Capability</td>
<td>FAR 25 The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List: 747-34-008-MDL Issue 0 dated 12 Jun; 12 later revisions of the above listed documents approved by EASA</td>
<td>Master Data List: 747-34-008-MDL Issue 0 dated 12 Jun; 12 later revisions of the above listed documents approved by EASA</td>
<td>CDPDLC system not compliant with Commission Regulation (EC) 296/2009</td>
<td>04/09/2012</td>
<td>Active</td>
<td>10041282</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SR02405LA</td>
<td>407</td>
<td>Survivor Window Raft Kit P/N 407-587-000 for Bell Helicopter Textron Model 407 Helicopters. Reference FAA STC SR02405LA.</td>
<td>The Certification Basis for the original product remains applicable. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>B407-Flight Manual Supplement to the Bell 407 Helicopter Flight Manual when equipped with the Survivor Window Raft Kit. Document No: AA-10078 Rev. NR, dated December 6, 2011, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/414/CF.</td>
<td>None</td>
<td>04/09/2012</td>
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<td>10041283</td>
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<td>AILIFTS A/S</td>
<td>REV. 1</td>
<td>MBB-BK117 C-2, MBB-BK117 D-3, MBB-BK 117 C-2</td>
<td>Initial issueNight Vision Imaging System (NVIS) certification in order to enable Night Vision Goggles (NVG) operation on MBB-BK117 C-2 helicopters. Revision 1 NVIS certification on MBB-BK117 D-3 helicopters with configuration described in the Configuration Control Document CC2-145-33-101-100; Revision 3...</td>
<td>The Type and OSD Certification Bases (OB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Initial issueAIlift A/S Flight Manual Supplement (FMS): FMS-145-33-101-01; Rev. 1 23rd April 2012; AIlift A/S Modifications of the Aircraft in accordance with FAA STC 10040513 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/414/CF.</td>
<td>Initial issueRefer to applicable FMS for list of equipment mandatory for NVIS certification. This Major Change is applicable only to helicopters delivered after NVIS approval.</td>
<td>11/03/2024</td>
<td>Active</td>
<td>10041311</td>
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<tr>
<td>ARMSTRONG AEROSPACE Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST03086CH</td>
<td>B747-400</td>
<td>Installation and activation of a Panasonic Global Communication Suite (GCS) System (+,-) including the Panasonic eXPhone system in accordance with Armstrong Aerospace Master Data List Doc. No: DC 104-9034-00; Revision H, dated December 14, 2012. The addition (installation) and activation of the eXPhone system...</td>
<td>The Certification Basis for the original product as amended by Armstrong Aerospace Master Data List Doc. No: DC 104-9034-00; Revision H, dated December 14, 2012. The following requirements: (b) For the areas of change: the following requirements...</td>
<td>Certification Plan E11042-250-001, Rev. 0 (initial STC) Certification Plan E11042-250-001, Rev. 0 (initial STC) Armstrong Aerospace Master Data List: DC104-9034-00; Revision H, dated December 14, 2012. The following requirements: (b) For the areas of change: the following requirements...</td>
<td>Modification of the aircraft in accordance with this STC requires prior concurrence modifications in accordance with FAA STC 10040513 (FAAAST ST03047CH) and 10040504 (FAA STC ST03086CH). FAA approved Aircraft Flight Manual Supplement DC303-9034-00 Revision A (Configuration A) or DC303-9034-10.</td>
<td>10/04/2013</td>
<td>Active</td>
<td>10041316</td>
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## List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<tbody>
<tr>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>10041328</td>
<td>FAA STC ST02961CH</td>
<td>B737-400/-700-800/-900ER</td>
<td>Validation of FAA STC ST02961CH - Approved configurations are listed following. Configuration 1: Replacement of Honeywell Weather Radar System RDR-480 with Honeywell Weather Radar RDR-4000 (+) Configuration 2: Software upgrade to Honeywell WRS RDR-4000 Existing Weather Radar Processor will be...</td>
<td>The Certification Basis for the original product remains applicable to this Supplemental Type Certification approval. The certified noise/ or emissions levels of the original product are unchanged and remain applicable to this Certification basis.</td>
<td>Configurations EASA STC ST02961CH-Master Data List ECS-208688, Revision Q, dated 23 January 2012; FAA STC ST02961CH, Revision Q, dated 24 January 2012; Transport Canada STC No. A23-15-10335 Issue A; or later revisions of the above listed documents approved by EASA.</td>
<td>07/02/2012</td>
<td>Active</td>
<td>10041328</td>
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<tr>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>10041337</td>
<td>REV. 1</td>
<td>MYSTERE FALCON 20</td>
<td>Installation, Aircraft Completion</td>
<td>Installation of Iridium Satcom System ICS-1206 applicable to DassaultAviation MF900 aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this Supplemental Type Certification approval. The requirements for environmental protection and the associated certified noise or emissions levels of the original product are unchanged and remain applicable to this Supplemental Type Certification approval.</td>
<td>Description as per MDL ref. MDL-006-11 Rev. 2, dated 21/08/2012; AFMS ref. UAC-AM-11-003 Rev. 2, dated 21/08/2012; Note: document reference UAC-CD-E-1004-11 Rev. 1 was modified on 09/09/2012 to record test results or later revisions of the above listed documents approved by EASA.</td>
<td>05/01/2015</td>
<td>Active</td>
<td>10041337</td>
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<tr>
<td>BOMBARDIER INC.</td>
<td>10041344</td>
<td>TCCA STC 5012-16</td>
<td>BD-700-1A10</td>
<td>Installation, Aircraft Completion Modifications - ASB 9456.</td>
<td>The Certification Basis for the original product remains applicable to this Supplemental Type Certification approval. The requirements for environmental protection and the associated certified noise or emissions levels of the original product are unchanged and remain applicable to this Supplemental Type Certification approval.</td>
<td>Transport Canada STC No. SA12-16 Issue 2 dated 20 March 2012; Bombardier Certification Configuration Definition Record RA2-BM700-001 Revision Y dated 31 July 2012; CBCA STC ST02961CH-Master Data List ECS-208688, Revision Q, dated 23 January 2012; FAA STC ST02961CH, Revision Q, dated 24 January 2012; Transport Canada STC No. A23-15-10335 Issue A; or later revisions of the above listed documents approved by EASA.</td>
<td>Restricted to ASB 9456. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane's cabin using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certification basis.</td>
<td>10/06/2012</td>
<td>Active</td>
<td>10041344</td>
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<tr>
<td>TC INTER-INFORMATICS A B.</td>
<td>10041362</td>
<td>REV. 1</td>
<td>737-800</td>
<td>D27 Cabin layout reconfiguration</td>
<td>The Certification Basis for the original product remains applicable to this Supplemental Type Certification approval. The requirements for environmental protection and the associated certified noise or emissions levels of the original product are unchanged and remain applicable to this Supplemental Type Certification approval.</td>
<td>Definition document, reference D27-D01 revision 5, dated 30 October 2012; Project Compliance Checklist, reference D27-D01 revisions 2, dated 30 October 2012; Declaration of Compliance, reference D27-D05 revision 5, dated 30 October 2012; ECM Supplement to D-27370-86N-IIS, revision 0, dated 10 September 2012.</td>
<td>Limitations/Applicability of this STC is limited to the aircraft serial number listed in Definition Document D27-D01. Conditions Prior to installation of this STC it must be determined that the interrelationship between this STC and any other previouly installed modification and/or repair will introduce...</td>
<td>05/02/2013</td>
<td>Active</td>
<td>10041362</td>
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<td>EASA STC No</td>
<td>STC Holder</td>
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<td>10041373</td>
<td>JANMCO AMERICA, INC.</td>
<td>FAA STC ST02107SE-D</td>
<td>B777-300</td>
<td>Validation of FAA STC ST02107SE-D related to Interior Reconfiguration.</td>
<td>The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval. Requirements at a later amendment level than specified in TCD/SEASA.IM.003 are listed below:§25.853(a)(8)(a)(v)(v), Addit. 25.116 - v...</td>
<td>Janmco America’s Master Drawing List (MDL) rev 8, dated November 3, 2011; Airplane Flight Manual Supplement, document AFM1724-1, Rev. 1, dated November 6, 2011; Supplemental Instruction for Continued Airworthiness (ICA), document number ICA1724-1, Revision 0, dated September 19, 2011.</td>
<td>Approval of this change in type design is applicable to the above model aircraft only.</td>
<td>11/02/2012</td>
<td>Active</td>
<td>10041373</td>
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<tr>
<td>10041380</td>
<td>INFLIGHT WARNING SYSTEMS, Inc.</td>
<td>FAA STC ST02300LA</td>
<td>B777-200, 200LR, 300, 300ER</td>
<td>Validation of FAA STC ST02300LA related to Installation of Fan Vibration Monitor Unit.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Drawing List 21770B, Revision D, dated May 21, 2010; Installation and Operational Specifications (IOS) for Continued Airworthiness 12705, Revision A, dated May 21, 2010; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>None.</td>
<td>12/09/2012</td>
<td>Active</td>
<td>10041380</td>
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<tr>
<td>10041414</td>
<td>BERINGER AERO</td>
<td>REV. 4</td>
<td>ASW 20 F, ASW 20 FL, ASW 20 CL, ASW 20 C, ASW 20 L, ASW 20 BL, ASW 20 C, ASW 20 B, ASW 20 BL, ASW 20 , ASW 20 B</td>
<td>Installation of main wheels and brake systems P/N: JA-01 Revision 1 and new wheel P/N: RF-002(B) with brake EA-006E, which superseded the P/N JA-01; Revision 2: Update with new wheel P/N: RF-019, which superseded the wheel P/N: RF-002(B); Revision 3: Update with new servicing instructions with...</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification file CR-MOD001-STD-005 REV 0, dated 30 Nov 2012; Installation and Instruction for Continued Airworthiness: ICA-001-STD-005 REV 0, dated 12 Jan 2012; or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and any other previously installed change/repair is acceptable under the EASA system.</td>
<td>29/02/2014</td>
<td>Active</td>
<td>10041414</td>
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<tr>
<td>10041447</td>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FAA STC ST02964NY-D</td>
<td>MF50</td>
<td>Installation of Crane Elevator “SmartSystem” Torsion Pressure Sensor (TPS) System Validation of FAA STC ST02964NY-D</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Dassault Falcon Jet-Wilmingtom Mater Drawing List, F005- D001-055 Revision 1R dated January 04, 2012; Instructions for Continued Airworthiness doc., N° F030- D001-155 Revision B dated December 13, 2011; FAA STC N° ST02964NY-D issue dated February 21, 2012; or later revisions of the above listed document(s)</td>
<td>None.</td>
<td>18/09/2012</td>
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<td>10041447</td>
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<td>STC Holder</td>
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<tr>
<td>AIRCRAFT STRUCTURES INTERNATIONAL CORP.</td>
<td>208, 208B</td>
<td>Installation of an internal Cargo Pod lighting system.</td>
<td>Design. Non-significant change. The Certification Basis for the original product and the following additional/alternative airworthiness requirements are applicable to product change. As specified in the original FAA STC No. SA0110545C dated 02-July-2012. Environmental Protection: The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>FAA approved installation data list document DL1819M02, rev. IR, dated 28 June 2012; FAA approved Compliance data list document DL1819M00, rev. IR, dated 28 June 2012. FAA approved Instruction for continued airworthiness 00210000-ICA rev.A, dated 8 May 2012 or later revisions of the above listed documents...</td>
<td>If the holder agrees to permit another person to use this certificate to modify the product, the holder shall give the other person written evidence of that permission.</td>
<td>19/09/2012</td>
<td>Active</td>
<td>10041470</td>
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<td>KMLEX AVIATION INC.</td>
<td>FALCON 50EX</td>
<td>Relocation of Radio Altimeter Converter/Validation of TCCA STC No. Q-LSA10-047/D issue dated November 04, 2010.</td>
<td>The Certification Basis for the original product remains applicable to the certification basis. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List MDL00158402 - Relocation of Radio Altimeter Converter on Falcon 50 aircraft S/N 278&lt;(&gt;,&lt;) revision NC dated November 04, 2010; TCCA STC No. Q-LSA10-047/D issue dated November 04, 2010.</td>
<td>Restricted to S/N 278</td>
<td>19/09/2012</td>
<td>Active</td>
<td>10041473</td>
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<tr>
<td>ACS-NAI Ltd.</td>
<td>PC-6B2-H4</td>
<td>Installation of Vapour Cycle Air Conditioner System.</td>
<td>US CAR 3 dated May 15, 1956 Amnd: 3-1 to 3-nd Special Conditions notified by the FAA to the Swiss FOCAThe Certification Basis for the original product remains applicable to the certification basis. Derogation to latest amendment (Part 21 A 101). The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement H404091 Revision N/C; Maintenance Manual Supplement H404071 Revision NC and H404072 Revision N/C; Wiring Diagram Manual Supplement H404073 Revision N/C; FAA approved installation data list document DL1819M02, rev. IR, dated 28 June 2012; FAA approved Compliance data list document DL1819M00, rev. IR, dated 28 June 2012. FAA approved Instruction for continued airworthiness 00210000-ICA rev.A, dated 8 May 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decisi...</td>
<td>Refer to the Airworthiness Limitations contained in the Airworthiness Approval.</td>
<td>04/10/2012</td>
<td>Active</td>
<td>10041474</td>
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<tr>
<td>KMLEX AVIATION INC.</td>
<td>FALCON 50EX</td>
<td>Replacement of Aft lavatory faucet adapter - Dassault Falcon 50E/50/aircraft MSN 278 Validation of TCCA STC No. Q-LSA10-048/D issue dated November 04, 2010.</td>
<td>The Certification Basis for the original product remains applicable to the certification basis. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List MDL00158402 - Replacement of Aft lavatory faucet adapter on Falcon 50 aircraft S/N 278&lt;(&gt;,&lt;) revision NC dated November 04, 2010; TCCA STC No. Q-LSA10-048/D issue dated November 04, 2010.</td>
<td>Restricted to S/N 278</td>
<td>19/09/2012</td>
<td>Active</td>
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<td>HAWKER BECKWITH LTD.</td>
<td>B200GT</td>
<td>Introduce Rockwell Collins TCAS 4000, Traffic Collision Avoidance System (TCAS) II System.</td>
<td>FAR-23, per TCDSThe Certification Basis for the original product remains applicable to the certification basis. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airworthiness Approval; Master Document List MDL00159002 - Approval of Traffic Collision Avoidance System (TCAS) II System.</td>
<td>STC does not provide compliance with Commission Regulation (EC)1332/2011 as regards to ACAS II equipment.</td>
<td>05/10/2012</td>
<td>Active</td>
<td>10041476</td>
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<td>EASA STC no.</td>
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<td>10041486</td>
<td>DUNLOP AIRCRAFT TYRES LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>ERJ 170-200 STD, ERJ 170-200 LR, ERJ 170-100 STD, ERJ 170-100 LR.</td>
<td>Dunlop Aircraft Tyres Limited P/N EZ228167 Radial Nose Landing Gear installation on ERJ 170.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Document List 170 PR5 MDL Rev IR dated 11 September 2012 or later approved revisions of the above listed documents approved by EASA.</td>
<td>It is not allowable to mix different tyre types (radial and bias) in the nose landing gear leg.</td>
<td>20/09/2012</td>
<td>Active</td>
<td>10041486</td>
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<td>10041532</td>
<td>ORCON CORPORATION</td>
<td>Validation of FAA STC ST01834LA related to Installation of InsulationBlanket</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Orcon Corporation’s Index List, Document No. F93005, Revision H, dated 23 April, 2012</td>
<td>None</td>
<td>25/02/2012</td>
<td>Active</td>
<td>10041532</td>
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<td>10041533</td>
<td>ATLANTIC AERO, Inc.</td>
<td>Installation of modified battery heat shield</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Atlantic Aero Instructions for Continued Airworthiness no 26493011 revision IR dated 15 September 2011</td>
<td>None</td>
<td>25/09/2012</td>
<td>Active</td>
<td>10041533</td>
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<tr>
<td>10041546</td>
<td>GLOBAL AIRWORKS, Inc.</td>
<td>Installation of a Global AirWorks Cockpit Door Surveillance System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Standard Aero Master Data List, ref : MDL250071, Rev F, dated 01February 2012</td>
<td>None</td>
<td>26/09/2012</td>
<td>Active</td>
<td>10041546</td>
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<td>10041560</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>Installation is limited to Serial Numbers 435, 440, 441, 445 and 446</td>
<td>Installation must be in accordance with Field Aviation Master Data List , Revision IR, Maintenance must be in accordance with Field Aviation Aircraft Maintenance Manual Supplement 25390000, Revision IR.</td>
<td>Hayman Corporation’s lncorp’s Continued Airworthiness, Document No.: MDL-9000, Revision H, dated 23 April, 2010</td>
<td>None</td>
<td>26/09/2012</td>
<td>Active</td>
<td>10041560</td>
<td></td>
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<tr>
<td>10041576</td>
<td>JET AVIATION AG, BASEL</td>
<td>Exchange to Universal Avionics FMS UNS-1Lw Out: Universal Avionics FMS UNS-1Lw system</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List Doc. No.: BSDYB004-004 Rev. C of 21.02.2013</td>
<td>S/N: 216, 224 The approval holder shall fulfil the obligations of Part 21, Point 21.A 109 or 21.A 451 (b) Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and or repair will introducea...</td>
<td>25/02/2013</td>
<td>Active</td>
<td>10041576</td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAAC C/D</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10041590</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>REV. 1</td>
<td>TCCA STC SA-12-07</td>
<td>DHC-8-106, DHC-8-102, DHC-8-103,</td>
<td>Installation of Protective Breathing Equipment (PBE)</td>
<td>TCCA STC SA-12-57, Issue 1</td>
<td>Installation must be in accordance with FAA STC 25680000, Revision IR, dated 05 September 2012. Maintenance must be in accordance with FAA Aircraft Certifications.</td>
<td>None</td>
<td>15/10/2012</td>
<td>Active</td>
</tr>
<tr>
<td>10041595</td>
<td>JET AVIATION AG, BASEL</td>
<td>REV. 1</td>
<td>B737-900ER</td>
<td>EMS SATCOM AMT-3800 Installation.</td>
<td>FAR 25.1059 The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAR 25, 25-1059: The Certification Basis for the original product remains applicable to this certificate/approval. Do not materially contribute to safety. The certificated noise and or emissions levels of the original product are unchanged. Remains applicable to this certificate/approval.</td>
<td>Instructions for Continued Airworthiness, BB73723265-06, Rho later revisions of the above listed documents approved by EASA.</td>
<td>18/01/2013</td>
<td>Active</td>
<td>10041595</td>
</tr>
<tr>
<td>10041598</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING 787-200/-300</td>
<td>Installation of Inflight Antenna.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>STC Twenty One Ltd Master Document List No. 32.1.MD-1090 Issue 3 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>01/10/2012</td>
<td>Active</td>
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<tr>
<td>10041603</td>
<td>ASI INNOVATION</td>
<td>REV. 1</td>
<td>EMB-120ER</td>
<td>&quot;P-RNAV Certification/Description as per documents in “Associated Technical Documentation”.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Applicability restricted to the following S/Ns: 012, 013, 027, 028.</td>
<td>01/10/2012</td>
<td>Active</td>
<td>10041603</td>
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<tr>
<td>10041604</td>
<td>JET AVIATION AG, BASEL</td>
<td>REV. 1</td>
<td>B737-900ER</td>
<td>VIP Cabin interior installation.</td>
<td>FAR 25, 25-1059: The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>Application for Continued Airworthiness, BB37560-150-01-01 Rev. 1, issued 18 April 2012.</td>
<td>05/11/2012</td>
<td>Active</td>
<td>10041604</td>
<td></td>
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<tr>
<td>10041623</td>
<td>DASSAULT FALCON SERVICE</td>
<td>REV. 1</td>
<td>F2000, F2000EX</td>
<td>WIFI activation on Falcon 2000, F2000EX aircraft.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>[DFS Change Approval Sheet DFS101601-C2S001A dated 07.08.2012]</td>
<td>F2000 and F2000EX aircraft equipped with WiFi equipment installation compliant with prerequisite role DT-003-12.</td>
<td>02/10/2012</td>
<td>Active</td>
<td>10041623</td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certificate basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10041628</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td></td>
<td>777-300ER</td>
<td>N0419 Cabin Interior Modifications on Jet Airways Boeing 777-300ER Aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>N0419-D067 Rev. A, dated Oct 2, 2012; Certification Summary for the Cabin Interior Modifications on Jet Airways Boeing 777-300ER Aircraft</td>
<td>None</td>
<td></td>
<td>03/10/2012</td>
<td>Active</td>
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<tr>
<td>10041637</td>
<td>EUROCOPTER CANADA Ltd.</td>
<td></td>
<td>TCCA STC SH12-10</td>
<td>EC130B4</td>
<td>Door handle installation (P/N 120-2000147) fits STC in the validation of TCCA STC SH12-2-10 (issue date - February 10, 2012)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Definition and installation ECL Master Drawing List - doc ref. MDL 10-015 Issue B Installation Procedure - RH Door Handle Installation - EC130B4 - doc ref. IP-ECL-127 Rev. 2</td>
<td>None</td>
<td></td>
<td>03/10/2012</td>
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<td>10041648</td>
<td>STAR AVIATION, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST0004AT</td>
<td>747-400</td>
<td>Installation of Cabin Wireless Local Area Network (CWLAN) in accordance with Star Aviation Inc. Master Data List R44-0175, Revision G, dated October 13, 2004, or later FAA approved revisions and Provisional/Installation of a Panasonic Avionics Corporation eXPhone system.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List R44-0175, Revision G, dated October 13, 2004+ (&gt;,-,), Master Data List R44-0013, Revision D, dated July 19, 2012+ (&gt;,-), &gt;AFM Supplement R44-0933, Revision A, dated October 18, 2004</td>
<td>None</td>
<td></td>
<td>23/10/2012</td>
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<tr>
<td>10041649</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td></td>
<td>FAA STC ST0152AT-D</td>
<td>GV AND GV-SP</td>
<td>Installation of APU Fire Bottle</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Installed in accordance with Gulfstream Aerospace Corporation Index Link/GCS14428001, Revision L, dated July 31, 2011 Operation in accordance with AFM GCS14428001, dated April 8, 1999 or later FAA approved revision required for fully operational configurations (non-spare). For model GV, ICA GCS14428001...</td>
<td>1- The installer must determine whether this design change is compatible with previously approved modifications - If the holder agrees to permit another person to alter the product, the holder must give the other person written evidence of that permission.</td>
<td></td>
<td>04/10/2012</td>
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<tr>
<td>10041655</td>
<td>AEROCONSEIL</td>
<td></td>
<td>777-200LR &amp; 777-300ER</td>
<td>0465-01 - eXPhone System Activation on 777-200LR &amp; 777-300ER</td>
<td>The Certification Basis for the original product and the following: additional or alternative avionics: requirements are applicable to this certificate/approval.</td>
<td>MDL: 0465-01-A-2103-R01 dtd. 3 Oct 2012 AFMS 0465-D1-A-2301-R01 dtd. 23 Jul 2012 AOA 0465-01-B-2211-R00 dtd. 23 Jul 2012</td>
<td>None</td>
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<td>04/10/2012</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10041674</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>FAQ STC ST02868CH-D</td>
<td>10041711</td>
<td>CL-600-2B16 (CL-604 VARIANT)</td>
<td>Installation of Drain Mast</td>
<td>EASA STC No. ST02868CH-D: JET Aviation St. Louis, Inc. Master Data List 41-260G001, Rev C dated 09 February 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>None</td>
<td>05/10/2012</td>
<td>Active</td>
<td>10041674</td>
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<tr>
<td>10041706</td>
<td>SKYTRAC SYSTEMS LTD.</td>
<td>TCCA STC 5H10-51</td>
<td>10041712</td>
<td>A9139</td>
<td>Installation of a SkyTrac ISAT-100/100A Position Data Communicator (Canadian STC 5H10-51 issue 2)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/lightweight noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>09/10/2012</td>
<td>Active</td>
<td>10041706</td>
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<tr>
<td>10041707</td>
<td>BERINGER AERO</td>
<td>REV. 4</td>
<td>10041713</td>
<td>VENTUS A, VENTUS 2A DISCUS A, DISCUS 2A 1, DISCUS A, DISCUS 2A</td>
<td>Installation of main wheels and brake system P/N: JA-02 Revision 1: Update with new wheel P/N: RF-007B with brake EA-0064 which supersedes the PI-116; JA-02 Revision 2: Update with new wheel P/N: RF-020 which supersedes the wheel P/N: RF-007 Revision 3: Update with new servicing instructions within ICA.</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/lightweight noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification File: CR-MOD001-STM-067 REV 0, dated 30 Nov. 2023</td>
<td>01/03/2024</td>
<td>Active</td>
<td>10041707</td>
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<tr>
<td>10041711</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAQ STC NO. ST01541W-D</td>
<td>10041715</td>
<td>CL600-2B19 REGIONAL JET A21EA</td>
<td>Installation of Rockwell Collins FMS-4000 to provide WAAS/LP capability.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/lightweight noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC NO. ST01541W-D: 10 April 2009 Master Drawing List RIA-0179-001, Revision C, dated 11 October 2008</td>
<td>12/10/2012</td>
<td>Active</td>
<td>10041711</td>
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<tr>
<td>10041712</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAQ STC NO. ST01541W-D</td>
<td>10041717</td>
<td>CL-600-2B16 (CL-604 VARIANT)</td>
<td>Installation of Rockwell Collins FMS-6000 to provide SBAS LPV capability.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/lightweight noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC NO. ST01561W-D: 05 November 2016 Master Drawing List RIA-0165-001, Revision H, dated 1 October 2016</td>
<td>09/10/2012</td>
<td>Active</td>
<td>10041712</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FALCON 2000, FALCON 2000EX</td>
<td>Validation of FAA STC ST02962NY - Installation of ADS-B Out System on aircraft equipped with Rockwell Collins Pro Line 4 or 4 Pro Line 21 series avionics system Revision 1: Installation of ADS-B Out System compliant with DO-2008 Revision 2 - Correction of FAA STC number. Revision 3: Introduction of EASA STC number.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/approval.</td>
<td>-AFMS ref. F5XW-D0006-150 initial revision. dated 10 October 2011</td>
<td>-AFMS ref. F5XW-D0006-150, revision A, dated 21 December 2015</td>
<td>10/02/2019</td>
<td>Active</td>
<td>10041722</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>MYSTERE FALCON 50</td>
<td>Validation of FAA STC ST02962NY - Installation of ADS-B Out System in Dassault Aviation Mystere Falcon 50 aircraft equipped with Rockwell Collins Pro Line 4 or Pro Line 21Avionics System Revision 1: Installation of ADS-B Out System compliant with DO-2008 Revision 2: Correction of FAA STC number.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/approval.</td>
<td>-AFMS ref. F5XW-D0006-150 Rev. A, dated 20 April 2013;</td>
<td>-AFMS ref. F5XW-D0006-150 Rev. A, dated 14 October 2015</td>
<td>10/02/2019</td>
<td>Active</td>
<td>10041723</td>
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<td>EUROCOPTER CANADA Ltd</td>
<td>TCCA STC SH11-35</td>
<td>Glare Shield Modification P/N 734-85114 This STC is the validation of TCCA STC SH11-35 (issue No. 1 dated 29 June 2017).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/approval.</td>
<td>Operation Eurocopter Canada Limited Flight Manual Supplement - Adjustable Glare Shield Mod EC130 - doc. ref. FM9-EC1.14Rev 0</td>
<td>A, dated 30 Sept...</td>
<td>11/10/2012</td>
<td>Active</td>
<td>10041736</td>
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<tr>
<td>MID-CANADA MOD CENTER</td>
<td>TCCA STC O-LSA09-346D</td>
<td>TrueNorth Saltohphone Installation</td>
<td></td>
<td>Master Data List ref. F5XW-D0006-150 initial revision, dated 14 October 2015; -AFMS ref. F5XW-D0006-150, revision A, dated 14 October 2015</td>
<td></td>
<td>Restricted to ASN 8802.EASA STC Nos.10028546 and 10028538 Rev. 2 (Transport Canada STC Nos. O-LSA09-346D and C-LSA09-346D) are pre-requisites to this approval.</td>
<td>12/10/2012</td>
<td>Active</td>
<td>10041743</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FALCON 2000, FALCON 2000EX</td>
<td>Installation of provisions for navair Electronic Flight Bag. Rev. 1: Model 737-8 and additional configurations are included.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/approval.</td>
<td>-Master Data List ref. 0106-06191-2169 Revision AD dated 31 July 2017; -Instructions for Continued Airworthiness/Wiring Interconnect System/Source Document Ref. 07/06-06884-2169 Rev. F, dated 16 August 2017; -Airplane Flight Manual, Rev. A06-006837-2169 Rev. A, dated 29 June 2017</td>
<td>Installation of Nav-Arrow Mount &quot;Configuration 1&quot; (as per Master Data List) is not covered by the EASA validation. This STC does not approve the use of applications or approve installation of the navair display or docking station. Prior to installation of this design chang...</td>
<td>07/12/2017</td>
<td>Active</td>
<td>10041776</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>ASI INNOVATION</td>
<td>10041778</td>
<td></td>
<td></td>
<td>EMB-135LR</td>
<td>Cabin interior modification. Description as per Associated Technical Documentation.</td>
<td>Certification Basis for the original product remains applicable to this certification/ approval. The requirements for environmental protection and the associated/certificated noise emissions levels of the original product are unchanged and remain applicable to this certification/ approval.</td>
<td>ASI SDL, STC DMA List ref. ASI-SDL-0059-01, dated 12/10/2012; ASI MDL, Master Drawing List ref. ASI-MDL-0059-02, dated 01/10/2012; ASI Technical Dossier ref. ASI-TD-0059-01, dated 09/10/2012; ASI AFM ref. ASI-AFM-0059 issue 0, dated 27/09/2012 or later revisions of the above listed documents approved.</td>
<td>Applicability limited to EMB-135LR S/N 886.</td>
<td>15/10/2012</td>
<td>Active</td>
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<tr>
<td>NOUVELLE MANUFACTURE D'AEROSTATS</td>
<td>10041796</td>
<td>REV. 1</td>
<td>T-SERIES N-SERIES M-SERIES</td>
<td>B777-200LR</td>
<td>Transfer of STC 10041796_Nouvelle Manufacture d'Aérostat/Description of Design Change Modification + Diners à bord n, 2-guests kit and 16-guests kit</td>
<td>Certification Basis (CB) for the original product remains applicable to this certification/ approval.</td>
<td>Supplement Flight Manual NMA 01-2012 or later revisions of the above listed documents approved by EASA.</td>
<td>03/04/2017</td>
<td>Active</td>
<td>10041796</td>
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<tr>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>10041806</td>
<td>FAA STC ST018458E-D</td>
<td></td>
<td>747-400</td>
<td>Installation of a galley modification on Boeing 747-400 series aircraft in accordance with Jamco America, Inc. Master Drawing List (MDL) 1419-1 Rev.E, dated May 25, 2012, or later FAA approved revision. Validation of FAA STC ST018458E-D.</td>
<td>Certification Basis for the original product and the following/additional or alternative avionics/environmental requirements are applicable to this certification/approval.</td>
<td>Master Data List doc. N° MDL-0059-2500 Rev. C dated 16.10.2012 or later revisions of the above listed documents approved by EASA.</td>
<td>S/N 40753</td>
<td>16/10/2012</td>
<td>Active</td>
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<tr>
<td>JAMCO AMERICA, INC.</td>
<td>10041824</td>
<td>FAA STC ST02944CH</td>
<td></td>
<td>777-300ER, 777F, 777-200, 777-200LR, 777-300</td>
<td>Installation of an AC-DC Converter</td>
<td>Certification Basis for the original product and the following/additional or alternative avionics/environmental requirements are applicable to this certification/approval.</td>
<td>Jamco America, Inc. Master Drawing List 1419-1, Rev.E, dated May 25, 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/014/CF (or subsequent revisions of this decision).</td>
<td>S/N 29262</td>
<td>17/10/2012</td>
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<td>STC Holder</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>PHOENIX AEROSPACE LIMITED</td>
<td>A139</td>
<td>Introduction of SX16 Searchlight System</td>
<td>CS-29 Initial Issue</td>
<td>Flight Manual Supplement; PADS/FSM/139/002 Rev A; or later revisions of the above listed documents approved by EASA.</td>
<td>none</td>
<td>11/12/2012</td>
<td>Active</td>
<td>10041855</td>
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<td>PHOENIX AEROSPACE LIMITED</td>
<td>AW 139</td>
<td>Introduction of an electronic message sign</td>
<td>Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the aircraft; or emissions levels of the original product remain applicable to this certificate/support.</td>
<td>Flight manual supplement; PADS/FSM/139/001 Rev A; or later revisions of the above listed documents approved by EASA Master draw. List; PADS/MNU/1193 Rev. CICA; PADS/ICAA/1193 Rev. A &amp; PADS/MMS/1193/001 Rev. B</td>
<td>None.</td>
<td>11/12/2012</td>
<td>Active</td>
<td>10041863</td>
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<td>EUROCOPTER DEUTSCHLAND GmbH</td>
<td>EC 120 B</td>
<td>Installation of GTN 650 Unit on EC 120B</td>
<td>Certification Basis for the original product remains applicable to this certificate/support.</td>
<td>- SK-ARC-133-SUP-P-623 Flight Manual Supplement; Revision B; dated08-2012 or later revisions of the above listed documents approved by EASA- P547-001.0000-00 GTN Installation [Top Drawing]- SK-MKC-133-EO-P-616 Engineering Order; SK-ARC-133-CM-P-620 Customization Manual</td>
<td>none</td>
<td>23/10/2012</td>
<td>Active</td>
<td>10041896</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>GV</td>
<td>Installation of Honeywell TCAS 2000 TCAS II System</td>
<td>Certification Basis for the original product remains applicable to this certificate/support.</td>
<td>- FAA STC ST00673LA-D [&gt;] issued 17th May 1999-installed in accordance with Gulfstream Aerospace Corporation Index List ref. GC51 1298001, Revision C, dated 26th April 1999-Airplane Flight Manual Supplement, ref. GC51 1298001, Revision C, dated 26th April 2000 or later revisions of the above listed do...</td>
<td>None</td>
<td>23/10/2012</td>
<td>Active</td>
<td>10041900</td>
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<td>BERINGER AERO</td>
<td>R300/160 S, R300/180 R300/140, R300/160 R300/120, R300/120D R300/100 R2100, R2100, R2101, R2101, R2112, R2120 R1180T DR400-1180D/1 DR400-1180D/2 DR400-1180D/3 DR400-1180D/4 DR400-1250-1250/1 DR400-1250-1250/2 DR400-1200-1200/1 DR400-1200-1200/2 DR400-1200-1200/3 DR400-1200-1200/4 DR300/180R</td>
<td>Replacement of brake unitRevision 1: Addition of R 1000 series, R2000 series and R3000 series aircraft modelRevision 2:Addition of DR 300/180RRevision 3: Update with new servicing instructions within ICA document</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Refer to CS 29 paragraphs listed on document DM-STC-006(8); The requirements for environmental protection and the associated certificated noise and/or emissions levels of the orig...</td>
<td>Modification file: DM-STC-006 rev 09 dated 11 Mar. 2022; Installation and Instructions for Continued Airworthiness; ICA- STC-006(8) 01 dated 22 Oct. 2021 or later revisions of the above listed documents approved by EASA.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 21.A.109 or 21.A.451(b) (as applicable) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will intro...</td>
<td>18/03/2022</td>
<td>Active</td>
<td>10041955</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>A119, A119 MK II</td>
<td>STC for installation of Apical Industries, Inc. Cable Cutter Kit.</td>
<td>The Certification Basis for the original product remains</td>
<td>Apical Industries Inc. Master Document List</td>
<td>None</td>
<td>26/10/2012</td>
<td>Active</td>
<td>10041959</td>
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<tr>
<td>Wipaire Inc.</td>
<td>Installation of Wipaire Model 8000 Amphibian/Seaplane floats or WipaireModel 8750 Amphibian/Seaplane floats</td>
<td>The Certification Basis for the original product remains</td>
<td>Wipaire Inc. Installation Drawing as listed on the FAA STC No.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadworn effect upon the airworthiness of the product.</td>
<td>05/03/2019</td>
<td>Active</td>
<td>10041960</td>
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<tr>
<td>MB BK 117 C-2</td>
<td>Installation of a TETRA Digital Radio System into a NVIS certified Cockpit.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the product approval. CR1 on NVIS dated 11.09.2012 the requirements for environmental protection and the associated certification noise and/or emissions levels of the original pro...</td>
<td>- FMA 11-5-1 APPENDIX FOR TETRA Communication System rev. A dated 27.08.2012 or later revisions of the above listed documents approved by EASA. - Instructions for Continued Airworthiness Doc. No.: CA-117-23-10-0005 Rev. 0 or later approved revisions of the above listed documents.</td>
<td>Required modifications: EASA approval FAA 10040557 (ECmajor change E-1807) This aircraft employs Night Vision Imaging System lighting, commonly named NVIS lighting, that has been certified to ensure the aircraft is compatible with night vision goggles (NVG). For this reason modifications that are specific to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadworn effect upon the airworthiness of the product.</td>
<td>29/10/2012</td>
<td>Active</td>
<td>10041970</td>
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<tr>
<td>HA130 and HA130 D</td>
<td>Installation of S-TEC HeliSAS Stability augmentation system interfaced with PDA.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): EASA 0010014645 CRI InfPol 2709 issue 3 - EASA 0010014646 CRI CS27.1309 (FAA Special Condition 27-005-SC) For the installation in AS350B1, AS350B2</td>
<td>Rotorcraft Flight Manual Supplement, doc. ref. SA119-135/05-EC130/Rev. C, dated 30 April, 2015. For Airbus Helicopters type models AS350B1, AS350B2</td>
<td>IMC flight is not authorized External load operations are not authorized and the Helicopter is not compatible with the Garmin 500 HSI system. Aileron mix must be equipped with GDU 50V7 or later to output label 114 as True Track not Magnetic. Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadworn effect upon the airworthiness of the product.</td>
<td>10/08/2015</td>
<td>Active</td>
<td>10041971</td>
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<td>R 90-230RD</td>
<td>Avionic upgrade removed - Original flight instrument including KCS55 pictorial navigation system, attitude indication (with flight director), electrical attitude indication, encoding Attitude 2s, mechanical AS, VS and turn and bank. - Original vacuum system - Existing navigation system - Existing ATC tre...</td>
<td>FAR 23, AMDT 35The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the product approval. SCs or ESFs: For this change compliance to FAR 23, AMDT 35 and FAR 23.661 AMDT 36, 23.1306 AMDT 61, 23.1308 AMDT 57, 23.13...</td>
<td>Master Document List: MDL-201110675 Rev. B, dated 26.10.2012, Instructions for continued airworthiness: ICA-201110675 Rev. N/C, dated 27.06.2012, Rotorcraft Flight Manual Supplement for - Garmin 500 PFD/MFD: FMS-201110675.1 Rev. 0, dated 18.09.2012 - Garmin GTN750/560</td>
<td>None.</td>
<td>29/10/2012</td>
<td>Active</td>
<td>10041972</td>
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<td>STC Holder</td>
<td>STC No.</td>
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<td>Description</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>DERAIRFIND</td>
<td>FAA STC</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged/remain applicable to this certificate/approval.</td>
<td>FAA STC ST01566G</td>
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<td>GZH AEROSPACE (IRELAND) LIMITED</td>
<td>767-300</td>
<td>Installation of Teledyne Digital Flight Data Acquisition Unit</td>
<td>FAA STC ST01096CH</td>
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<td>BOMBARDIER INC</td>
<td>STC-AT-054-RevB</td>
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<td>10041987</td>
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<td>AIRBORNE MAINTENANCE AND ENGINEERING SERVICES, Inc.</td>
<td>STC-AT-054, Rev.3 dated 04.05.2015</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Certification basis</td>
<td>Description</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>DERAIRFIND</td>
<td>FAA STC</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged/remain applicable to this approval.</td>
<td>FAA STC ST01566G</td>
<td>AMES Master Drawing List RPT-69-06040, Revision G, 22/04/2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/404/CF (or subsequent revisions/offs decision)</td>
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<td>AIRBORNE MAINTENANCE AND ENGINEERING SERVICES, Inc.</td>
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<td>GZH AEROSPACE (IRELAND) LIMITED</td>
<td>767-300</td>
<td>Installation of Electronic Standby Instrument System (ESIS)</td>
<td>FAA STC ST03235CH</td>
<td>AXB Air Master Data List RPT-06-07307, Revision B, 16/08/2006 AFM Supplement, ABX Air Inc. Document No. RPT-66-07316, Revision IR+[+]/- 09/09/2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/404/CF (or subsequent revisions/offs decision)</td>
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<td>BOMBARDIER INC</td>
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<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>EC 655 T2</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any int....</td>
<td>EC 655 T2+ EC 655 P2+, EC 635 T1, EC 135 T2; EC 135 T1, EC 135 T2; EC 135 P2; EC 135 P1</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
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<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>EC 655 T2</td>
<td>Supplement to the AFM - Provisions for surveillance operation. Doc. No. AFM-1 STC-AT-054-Rev. 5 dated 04/05/2015</td>
<td>EC 655 T2+ EC 655 P2+, EC 635 T1, EC 135 T2; EC 135 T1, EC 135 T2; EC 135 P2; EC 135 P1</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval without any int....</td>
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<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>EC 655 T2</td>
<td>Installation of Electronic Standby Instrument System (ESIS)</td>
<td>EC 655 T2+ EC 655 P2+, EC 635 T1, EC 135 T2; EC 135 T1, EC 135 T2; EC 135 P2; EC 135 P1</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval without any int....</td>
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<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>EC 655 T2</td>
<td>Installation of Electronic Standby Instrument System (ESIS)</td>
<td>EC 655 T2+ EC 655 P2+, EC 635 T1, EC 135 T2; EC 135 T1, EC 135 T2; EC 135 P2; EC 135 P1</td>
<td>Installation of Electronic Standby Instrument System (ESIS)</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10042012</td>
<td>NSI INDUSTRIES S.A.</td>
<td>REV. 1</td>
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<td>AS 350 B4/AS 350 B1/AS 350 B2 / AS 350 B3/AS 350 B4/AS 350 BB</td>
<td>Installation of Data Tracking System DTS-Alpha Modifications defined in STC Data List, doc. SDL-DTSALPHA-AS350 Rev.04Revision 1 is a re-print following a change of name and address.</td>
<td>The Certification Basis for the original product and the following: additional or alternative airworthiness requirements are applicable toots certificate/ approval. This modification complies with CS-27 Amdt. 2 requirements as specified Certification Plan PCERT-DTSALPHA-AS350 Rev.04, dated 29/10/20...</td>
<td>Definition and installation Kit Part List, doc. NIM-KIT-DTS-03-00 Ind. 00Definition Document DTS Alpha PIN 412805 Amdt. B, doc. DD-412805 Rev.05STC Data List, doc. SDL-DTSALPHA-AS350 Rev.04 DOTS Alpha Installation Manual on AS350, doc. IM-DTSALPHA-AS350 Rev.05, Maintenance, Installation and Operation...</td>
<td>Modification not approved for rotorcraft equipped with AFCS (Automatic Flight Control System) or Weather Radar. Modification not approved for AS350B3 model helicopters without modification OP-4300 (Arriel 20 engine installation) incorporated. Prior to installation of this modification it must be determined...</td>
<td>14/06/2013</td>
<td>Active</td>
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<td>10042013</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC NO. SAS031SW-D</td>
<td>58A 58</td>
<td>Installation of S-TEC system 45/50 model ST-234-40/50, according to bulletin n. 324 dated 22-Sep-1983 and Master Drawing List No. 92252, dated 22-Sep-1983 (28 - System).</td>
<td>Design Non-significant change. The Certification Basis of the original product remains applicable toots certificate/ approval. Environmental Protection: The certified noise and/or emissions levels of the original product unchangeable and remain applicable to this certificate/approval.</td>
<td>FAA approved S-TEC Master Drawing List No. 92252, dated 22-Sep-1983 (28 - System), FAA / DAS Approved Pilot's Operating Handbook and/or Aircraft Flight Manual Supplement No. 83950 dated 5-Apr-1984 for system 40 FAA / DAS Approved Pilot's Operating Handbook and/or Aircraft Flight Manual Supplement No. 8, FAA / DAS Approved Pilot's Operating Handbook and/or Aircraft Flight Manual Supplement No. 8...</td>
<td>31/10/2012</td>
<td>Active</td>
<td>1042013</td>
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<td>10042023</td>
<td>MECRAER AVIATION GROUP S.p.A.</td>
<td>A1W139</td>
<td>Ballast Installation P/N 6AB1AD018-001</td>
<td>The Certification Basis for the original product remains applicable toots certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchangeable and remain applicable to this certificate/approval.</td>
<td>FAA approved S-TEC Master Drawing List No. 92252, dated 22-Sep-1983 (28 - System), FAA / DAS Approved Pilot's Operating Handbook and/or Aircraft Flight Manual Supplement No. 83950 dated 5-Apr-1984 for system 40, FAA / DAS Approved Pilot's Operating Handbook and/or Aircraft Flight Manual Supplement No. 8, FAA / DAS Approved Pilot's Operating Handbook and/or Aircraft Flight Manual Supplement No. 8...</td>
<td>This STC is only installable on AW139 Helicopters equipped with AgustaWestland Active Vibration Control System PIN 451830F00111 or 451830F00211.</td>
<td>05/11/2012</td>
<td>Active</td>
<td>1042023</td>
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<td>10042037</td>
<td>FLYBE LIMITED</td>
<td>FAA STC NO. SAS031SW-D</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>CAA UK RESPONSIBILITY</td>
<td>Ballast Installation P/N 6AB1AD018-001</td>
<td>The Certification Basis for the original product remains applicable toots certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchangeable and remain applicable to this certificate/approval.</td>
<td>FAA approved S-TEC Master Drawing List No. 92252, dated 22-Sep-1983 (28 - System), FAA / DAS Approved Pilot's Operating Handbook and/or Aircraft Flight Manual Supplement No. 83950 dated 5-Apr-1984 for system 40, FAA / DAS Approved Pilot's Operating Handbook and/or Aircraft Flight Manual Supplement No. 8, FAA / DAS Approved Pilot's Operating Handbook and/or Aircraft Flight Manual Supplement No. 8...</td>
<td>08/11/2012</td>
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<td>10042039</td>
<td>ORCON CORPORATION</td>
<td>FAA STC NO. SAS031SW-D</td>
<td>CL600-2B19</td>
<td>Orcom Thermal/Acoustic Insulation System for Bombardier Challenger 850Series Aircraft</td>
<td>The Certification Basis for the original product remains applicable toots certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchangeable and remain applicable to this certificate/approval.</td>
<td>FAA STC No. ST02050LA dated 8th May 2011 Orcon Corporation's Index List, No. FS0006, Revision B, dated April 20, 2011. Orcon Corporation's Continued Airworthiness Instructions, Document No. CAI-CL850 Revision A, dated January 5, 2011 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>05/11/2012</td>
<td>Active</td>
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<td>10042045</td>
<td>ASI INNOVATION</td>
<td>ASI INNOVATION</td>
<td>REV. 2</td>
<td>E30</td>
<td>Master with relay antenna installation Revision 1 - Update 03/2001.</td>
<td>The Certification Basis for the original product remains applicable toots certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchangeable and remain applicable to this certificate/approval.</td>
<td>ASI-FAM-0101-03 dated 23/07/2013AEE-SDL-0103-01 dated 23/07/2013 AEE-AMT-0101-01 dated 24/05/2013 or later revisions of the above listed documents approved by EASA.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Paragraphs 22.17, 10 Prior to installation of this modification it must be determined whether the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect on the...</td>
<td>28/06/2013</td>
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<td>STC Holder</td>
<td>FAA STC</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>ARMSTRONG AEROSPACE Inc.</td>
<td>ST01997CH</td>
<td>Installation of Honeywell Resonix 404AF/AFN system on DC-9-83(MD-83). This Rev. 2 is to correct a typo in the original STC number.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC ST01997CH</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise and/or emissions into the operational environment.</td>
<td>10/02/2012</td>
<td>Active</td>
<td>10402075</td>
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<td>AEROLITE AG</td>
<td>AIR139</td>
<td>Installation of Honeywell Resonix 404AF/AFN system on DC-9-83(MD-83). This Rev. 2 is to correct a typo in the original STC number.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC AIR139</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise and/or emissions into the operational environment.</td>
<td>10/02/2012</td>
<td>Active</td>
<td>10402076</td>
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<tr>
<td>JAMCO AMERICA, INC.</td>
<td>ST01514SE</td>
<td>Installation of Lavatory and galley modification of 747-400 series airplanes inaccordance with Jamsco America, Inc. Master Data List IA-10600, Rev. 6 dated December 2006.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC ST01514SE</td>
<td>None.</td>
<td>07/11/2012</td>
<td>Active</td>
<td>10402091</td>
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<tr>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>ST02166SE</td>
<td>Installation of additional Crew Rest Bunks in the Door 5 Overhead Crewrest.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection, the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC ST02166SE</td>
<td>Maximum number of occupants of the Door 5 Overhead crewrest is eighters.</td>
<td>08/11/2012</td>
<td>Active</td>
<td>10402096</td>
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<tr>
<th>EASA STC No</th>
<th>STC Holder</th>
<th>STC Code</th>
<th>Rev. Level</th>
<th>Foreign/NAA STC Code</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification Basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
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<tbody>
<tr>
<td>10042096</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC SA0118WV</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td></td>
<td>Installation of the Concorde FLA or VRSLA RG series batteries in Cessna single Engine Airplanes listed on the attached Approved Model List (AML) in accordance with Concorde Battery Corporation, Master Drawing List Deg. No. 5-0148, Revision A, dated August 16, 2012, or later FAA Approved revisions. Validation...</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Concorde Battery Corporation, Master Drawing List Deg. No. 5-0148, Revision A, dated August 16, 2012. Concorde Battery Corporation, Instructions for Continuing Airworthiness for the RG series batteries Doc. No. 5-0117, Revision M, dated March 15, 2012.</td>
<td>None</td>
<td>08/11/2012</td>
<td>Active</td>
<td>10042096</td>
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<td>10042101</td>
<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>FAA STC ST02004AT</td>
<td>DHC-8-301,-311,-314,-315 DHC-8-201,-202 DHC-8-102,-103,-106</td>
<td>Installation of a Honeywell MK VIII Enhanced Ground Proximity Warning System (EGPWS).</td>
<td>FAR 25 as per EASA/M.A.191, Issue 6 (inc. EWS) The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Installation Must be in accordance with Chippewa Aerospace Inc. Master Drawing ListD8-3443-001 and Engineering Order BS-3443-1001-c-... Rev. IR, dated 28 October 2002. Operation Must be in accordance with approved Chippewa Aerospace Inc. Flight/Navigation Supplement D6-3443-01-1012, Rev. 5 dated 12 Mar.</td>
<td>The aircraft must be equipped with the factory installed Honeywell Ground Proximity Warning System (GPWS).</td>
<td>None</td>
<td>08/11/2012</td>
<td>Active</td>
<td>10042101</td>
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<tr>
<td>10042180</td>
<td>FOKKER SERVICES B.V.</td>
<td>B737-600/-700/-800/-900ER</td>
<td></td>
<td>Introduction of LED light assemblies in the Boeing B737 NG aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td></td>
<td>Certification Plan CP-0110, Issue 1 Certification Record Report CR0-0110, Issue 1CIA Supplement SE-755er later revision of the above listed documents approved by EASA</td>
<td>None</td>
<td>12/11/2012</td>
<td>Active</td>
<td>10042190</td>
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<tr>
<td>10042171</td>
<td>SAA SOLUTIONS FOR AVIATION, S.L.</td>
<td>C-212-200</td>
<td></td>
<td>Installation of Iridium satellite phone. A satellite phone is added to the airplane. This satellite phone is not located in the cockpit and it is used only by an operator located at the rear of the passenger cabin, according to “Plan de Certificación A-203-CPL01”, Edición 8, Date: 12/11/2012.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td></td>
<td>1. “Plan de Certificación A-203-CPL01”, Ed 9, Date: 12/11/2012, 2. “Suplemento al Manual de Vuelo A1203-HKD01”, Ed. 3, Date: 10/05/2012, 3. “Suplemento al Manual de Mantenimiento A1203-HKD02”, Ed. 2, Date: 07/05/2012, 4. “Suplemento al Manual de Cableado A1203-HKD03”, Edición 3, Date: 04/09/2012 or later.</td>
<td>None</td>
<td>12/11/2012</td>
<td>Active</td>
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<td>10042191</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST02049RY</td>
<td>Installation of Panasonic Avionics Corporation System 3000/Passenger/Entertainment System in accordance with Delta Engineering Master Drawing List D8-3443-001, Revision D, dated June 08, 2005 or later approved revisions. Validation of FAA STC ST02049RY.</td>
<td>Installation: Must be in accordance with approved and Engineering Order BS-3443-1001, Rev. 5 dated 28 October 2002. Operation: Must be in accordance with approved Chippewa Aerospace Inc. Flight/Navigation Supplement D6-3443-01-1012, Rev. 5 dated 12 Mar.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td></td>
<td>Pane de Certificación A-203-CPL01”, Ed 9, Date: 12/11/2012, 2. “Suplemento al Manual de Vuelo A1203-HKD01”, Ed. 3, Date: 10/05/2012, 3. “Suplemento al Manual de Mantenimiento A1203-HKD02”, Ed. 2, Date: 07/05/2012, 4. “Suplemento al Manual de Cableado A1203-HKD03”, Edición 3, Date: 04/09/2012 or later.</td>
<td>None</td>
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<td>10042194</td>
<td>AIRCRAFT CABIN MAINTENANCE GmbH</td>
<td>B747-400</td>
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<td>Installation of refurbished Passenger Seat Belt PIN EPA 6488-2-01-40 sn(16g) seats</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
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<td>Date of Issue</td>
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<td>System Record</td>
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<td>ARMSTRONG</td>
<td>A320-232, A321-232 A319-131, A319-133</td>
<td>EASA validation of FAA STC ST02377CH Installa...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for...</td>
<td>none</td>
<td>20/06/2013</td>
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<td>ARMSTRONG</td>
<td>A321-231, 232 A320-111, 211 A319-131, 132, 133 A319-111, 112, 113, 114, 115</td>
<td>EASA validation of FAA STC ST02314CH Installation of an Aft Cargo Bay Equipment Rack</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for...</td>
<td>none</td>
<td>15/11/2012</td>
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<td>PMAV ENGINEERING</td>
<td>B747-400</td>
<td>Installation of provisions for the use of EFB Class 2.</td>
<td>The Certification Basis for the original product and the follow...</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>747-400</td>
<td>Cons...</td>
<td>The Certification Basis for the original product...</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>CL-600-281 CL-600-141 CL-600-2A12</td>
<td>Flight Management System NZ-2000 - Upgrade to SW 6.1</td>
<td>As per TCD05The Certification Basis for the original product remains applicable to this certificate/ approval. The...</td>
<td>none</td>
<td>21/11/2012</td>
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<td>STC Holder</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>RUAG AEROSPACE SERVICES Gmbh</td>
<td>10042291</td>
<td>ACSS TCAS II / 2000 Change 7.1 - Lower Cabin Altitude Modification Certificate</td>
<td>As per TCDS: The Certification Basis for the original product appends applicable item to this certificate.</td>
<td>RUAG Master Document List133G-MLD Rev N/C dated 19 November 2012</td>
<td>None</td>
<td>21/11/2012</td>
<td>Active</td>
<td>10042297</td>
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<td>BIZJET INTERNATIONAL SALES AND SUPPInc.</td>
<td>10042292</td>
<td>Installation of a quick drain kit on Environmental Control Unit/Validation of FAA STC ST09033SC</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>FAA STC ST09033SC</td>
<td>None</td>
<td>21/11/2012</td>
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<td>THE BOEING COMPANY</td>
<td>10042295</td>
<td>Lower Cabin Altitude (LCA) modification: 11.090 ft at 11.100 ft, flight altitude</td>
<td>Lower Cabin Altitude modification D926A200 - Lower Cabin Altitude Modification Certificate - Initial i.</td>
<td>EASA IM A.S.01059, EASA IM A.S.03158</td>
<td>None</td>
<td>04/07/2013</td>
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<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>10042297</td>
<td>Installation of ACSS Terrain and Traffic Collision Avoidance System (T2CAS) on Aircraft</td>
<td>The Certification Basis (CB) for the original product appends applicable to this certificate/ approval.</td>
<td>ACSS Master Data List document, ref: 8900280-04, Revision C, dated 17 October 2014</td>
<td>None</td>
<td>10/06/2022</td>
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<td>ASSOCIATED AIR CENTER, L.P.</td>
<td>10042299</td>
<td>Validation of FAA STC ST09033SC - Amendment to VIP Interior/STC</td>
<td>The Certification Basis for the original product can be amended by adding additional or alternative aircraft noise/ emissions levels of the original product.</td>
<td>- FAA Supplemental Type Certificate ST10913SC, amended November 17, 2014 - Master Document List, reference document 29052000001 Rev. E, dated November 17, 2014</td>
<td>None</td>
<td>21/11/2014</td>
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<td>Description</td>
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<td>10042314</td>
<td>BOMBARDIER Inc.</td>
<td>STC Holder</td>
<td>REV. 5</td>
<td>TCCA SA12-16</td>
<td>BD-700-1A10, BD-700-1A11</td>
<td>Installation, Aircraft Completion Modifications – EASA IMA. 00980-B700-1A10 and BD-700-1A11</td>
<td>Revision 2 includes the installation of large glass items in the cabinappropriate to Aircraft S/N 60003/Revision 3 updates the certification basis for the extinguishersaccording to CS 25.851(a) and (c) at Amti 12Re...</td>
<td>TCCA STC SA12-16B</td>
<td>Bombardier Configuration Definition Record RAZ-BM700-001 Rev BM/Bombardier/AFMS G-FMS05440/01</td>
<td>None.</td>
<td>24/06/2021</td>
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<td>10042345</td>
<td>TAMARACK</td>
<td>FAA STC SAU2216SE</td>
<td></td>
<td>SR20/SR22</td>
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<td>Removal of mechanical link between engine throttle control and propeller RPM control. Installation of a separate RPM control lever. Linkage between Cirus RPM cabiceman follows, center console to accommodatea new lever configuration, new knobs for all levers, and newinstallation hardware/FAA STC SA...</td>
<td></td>
<td>None.</td>
<td>26/11/2012</td>
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<td>10042346</td>
<td>ACS-NAI Ltd.</td>
<td>TCCA SA02-127</td>
<td>DHC-8-301,-311,-314,-315 DHC-8-201,-200 DHC-8-102,-103,-106</td>
<td>Installation of an Improved Ground Power Current Limiter(ACS/NAI Ltd. Modification Data Summary D 11900)TCCA STC SA02-127, Issue 1</td>
<td></td>
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<td>10042353</td>
<td>NEXTANT AEROSPACE, L.L.C.</td>
<td>FAA STC STO2371A</td>
<td>400A</td>
<td>Installation of Williams International F244-3AP engines as replacementfor the original Pratt &amp; Whitney Canada JT15D-5 units and associated systems. Installation of Nextant Winglets(Configuration B).</td>
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<td>23/06/2017</td>
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<tr>
<td>10042362</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA SH 90-4 ISSUE 4</td>
<td>AS355F2, AS355N, AS355NP, AS355E, AS355F, AS355P1</td>
<td></td>
<td></td>
<td>Installation of rear locker extender as defined in Dart Aerospace LtdMaster Installation Instructions No. MDL-D350-604 Rev C/Note 1. This approval amends DGAC STC STC 568-S0949 to add the AS355Fand AS355NPaircraft models. Note 2. This STC is the validation of TCCA STC SH90-4 Issue 4.</td>
<td>The Certification Basis for the original product remains applicable tothis certificate/approval. The requirements for environmental protection and the associatedcertificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>FMS D350-604 Rear Locker Rev B dated 15/11/2011 or later revisions of the above listed document(s) approved by EASA in accordance with the Technical Implementation Procedures of EU-CanadaBilateral Agreement.</td>
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<td>27/11/2012</td>
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<td>STC Holder</td>
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<tr>
<td>SERAM SARL</td>
<td>Installation of a Garem GNS 430/460 AV system, a GA 35/56 GPS antenna, a CI 198A CDS indicator, a Bendix/King KT 155 NAV/COM system, a 216-49/99 VOR/ILS indicator, a KN 62A DME receiver, GPS annunciator lights and an A22V, KA60, CI 105 antennas. The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>AFM supplement FM060R for BRNAV/PRNAV/ RNPage operations revision 0 dated 02/10/2012; AFM supplement FM060R for IFR operations revision 0 dated 30/10/2012; AFM supplement FM060R for VFR operations, revision 0 dated 11/06/2012; SERAM modification approval sheet n° FM060R dated 26/11/2012 or later revision.</td>
<td>Initial installation of this design change must be performed by authorised SERAM technical personnel. This installation is approved for VFR, IFR, RNAV and RNP Approach operations without coupling of navigation systems to an autopilot or altimeter. IFR approaches are authorised only with DNS 430s...</td>
<td>None</td>
<td>27/11/2012</td>
<td>Active</td>
<td>10042374</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>Installation of EROS Convertible ML20 Series Crew Oxygen Mask - Regulator with Detachable Smoke Goggles STC revision of the EASA STC 10042374 to include GIV-X model. The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>FAA STC ST01751LA-D</td>
<td>- FAA STC ST01751LA-D, amended on 13th May 2014; - FAA Statement of Compliance, ref. 10004-14-179, dated 28th May 2014... Installed in accordance with Gulfstream Aerospace Corporation Index List, ref. GC544538300, Revision 0, dated 12th May 2014; - EASA/JAR Compliance Report, ref. CES28943005, Revision N...</td>
<td>None</td>
<td>30/05/2014</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>Installation of an Airshow RF/IF Remote Controller. The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAA STC ST01957CH-D</td>
<td>- FAA STC ST01957CH-D, re-issued 31st July 2012; Installed in accordance with Gulfstream Aerospace Corporation Master Drawing List GCS16208600, Rev. A, dated 9th September 2004; - EASA/JAR Compliance Report, ref. CES28943004, Revision - dated 12th May 2014; - EASA/JAR Compliance Report, ref. CES28943003, Revision - dated 17th November 2012; - Instruction for Continued Airworthiness, ref...</td>
<td>None</td>
<td>28/11/2012</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>Installation of a Trakka A800 Searchlight Installation in a GV-SP aircraft. The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAA STC ST02518CH-D</td>
<td>- FAA STC ST02518CH-D, re-issued 9th October 2012; Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List GC528043003, Revision A, dated 11/06/2012; SERAM installation in GV-SP, Serial Number 5171. This STC approval is limited to IFR operations using a flight director. LPV approaches are approved for VFR, IFR, RNAV and RNP Approach operations without coupling of navigation systems to an autopilot or altimeter.</td>
<td>None</td>
<td>29/11/2012</td>
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<td>10042407</td>
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<td>ESG (EASA,R.S.01577)</td>
<td>Trakka A800 Searchlight Installation in a GV-SP aircraft. The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>E135 T2, EC135 T2+, EC135 P2+, EC135 T1, EC135 P1, EC135 P2: Trakka A800 Searchlight Installation in a GV-SP aircraft. The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>EC135 Flight Manual Supplement &quot;Trakka A800 Searchlight Installation&quot;0010-FMS-070, Rev. 01. The above listed documents are approved by EASA for Continued Airworthiness (AMM, HDM, IPC Supplements)0010-ICA-081, Rev. 01. Installation Instruction - ESG Modification 10042409</td>
<td>None</td>
<td>29/11/2012</td>
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<td>STC Holder</td>
<td>DG-500 M</td>
<td>Models</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>JET AVIATION, AG, BASEL</td>
<td>REV. 1</td>
<td>A319-115</td>
<td>Change to VIP Cabin Interior Outfitting</td>
<td>This change to STC 10042418 allows dual operation Private or Commercial Conversion from Private to Commercial Operation and from Commercial to Private. Requires a maintenance action See Master Data ListB/J04353-001 Rev. E</td>
<td>- Declaration of Compliance ADC 3491 Rev. IR, dated 28/11/2012; Compliance Record List CRL23-12/31M Issue 1 Rev. 0; dated 27/11/2012; Supplemental Type Definition Document ST0203-12/31M Issue 1 Rev. 0; dated 28/11/2012; Airplane Flight Manual Supplement A-00-1245-APM Issue 1 Rev. 1, dated 11/12/2009.</td>
<td>29/11/2013</td>
<td>Active</td>
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<td>APPAROE SYSTEMS, LLC.</td>
<td>REV. 1</td>
<td>AS 350 B3, AS 350 BA AS 350 B1, AS 350 B2</td>
<td>Installation of Vision 1000 System in accordance with Vision 1000 Master Document List 606586-000011 This STC is the validation of FAA STC SR0297Ch (date of issuance December 28, 2009)</td>
<td>- Approval of Vision 1000 System in accordance with Vision 1000 Master Document List 606586-000011, Revision 1.0 dated 27 April, 2013.</td>
<td>None</td>
<td>None</td>
<td>30/11/2012</td>
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<td>10042428</td>
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<td>EADS-CASA</td>
<td>REV. 1</td>
<td>A330-243</td>
<td>Installation of a Cockpit Door Surveillance System in the A330</td>
<td>- Approval of Cockpit Door Surveillance System in the A330</td>
<td>None</td>
<td>None</td>
<td>30/11/2012</td>
<td>Active</td>
<td>10042436</td>
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<td>BINDER MOTORENBAU GmbH</td>
<td>REV. 1</td>
<td>DG-500 M</td>
<td>TM-Triebwerk-B3 - Improved Engine Mounting</td>
<td>Exchange of the engine mounting plate from an aluminum version</td>
<td>- Technical Mitteilung (TM Triebwerk- B3) Solo- Umbau – OF-Entlallateur</td>
<td>None</td>
<td>25/10/2019</td>
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<td>AEROSPACE DESIGN AND CERTIFICATION Ltd. TIA AERODAC</td>
<td>REV. 1</td>
<td>A340-300</td>
<td>Installation of new Business Class Avio EB22 Seats and IMS Rape IFE System. The FWD cabin is currently configured with B/Cab E1 seats which will be removed and replaced with new Avio BC seats. The Rockwell Collins IFE System will be removed from the FWD Cabin and replaced with the IFS/ IFE system.</td>
<td>- The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>None</td>
<td>None</td>
<td>03/12/2012</td>
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<td>KUERZI AVIONICS AG</td>
<td>REV.1</td>
<td></td>
<td>PC-12/45</td>
<td>Additional Functions to STC 10042467- New Universal Avionics software Release EFI800 SON 107.7.1 installed. TAWS Alert installed - TAWS Speaker Alerts installed. ADIS-8 &quot;Out&quot; inhibit switch installed. For this change compliance to FAR 23 and FAR 23.62 (d) Amrd 44 and FAR 23.571(c) Amrd. 48, FAR 23.573(b).</td>
<td>STC holder shall fulfill the obligations of Part 21 § 21.118 Prior to installation of this modification it must be determined that the interaction between this modification and any other previously installed modification and/or repair will introduce no adverse effect(s).</td>
<td>Master Document List 1: MLD-201110653 Rev. A dated 19.06.2014 Master Document List 2: MLD-201314477 Rev. N.C. dated 19.06.2014.</td>
<td>S/Ns. 134; HB-FOG.</td>
<td>23/06/2014</td>
<td>Active</td>
<td>10042467</td>
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<td>OGMA INDUSTRIA AERONAUTICA</td>
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<td>EMB-145LU</td>
<td>VIP room extension and Iridium Satellite system installation on EMB-145</td>
<td>The change is applicable to the following aircraft having OGMA STC 10015799 incorporated. and that do not have flux valve sensors: TVA SN145076, 145084, 145129, 145135, 145147.</td>
<td>- Change Document No. 12P06-000 V0.1 dated 25-Sep-2012.</td>
<td>05/12/2012</td>
<td>Active</td>
<td>10042497</td>
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<td>AIR ALLIANCE GmbH</td>
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<td>PA-44-180</td>
<td>Replacement of Installed Gyro System by Aspen EFI 1000 Pro Primary Flight Display.</td>
<td>If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.</td>
<td>- Master Data List MDL 003 Rev. 13 dated 30.11.2012.</td>
<td>06/12/2012</td>
<td>Active</td>
<td>10042498</td>
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<td>AIR ALLIANCE GmbH</td>
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<td>PA-44-180T</td>
<td>Replacement of Installed Gyro System by Aspen EFI 1000 Pro Primary Flight Display.</td>
<td>- Airplane Flight Manual Supplement. 2012-014 PA44_506 Rev. 0.</td>
<td>- Airplane Flight Manual Supplement 2012-014 PA44_506 Rev. 0.</td>
<td>- Master Data List 2012-014 PA44_Form 2 Rev. 0.</td>
<td>05/12/2012</td>
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<td>EASA STC No.</td>
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| 10042513    | ASI INNOVATION             | REV. 4|                    | ATR 72-201, ATR 42-300, ATR 42-500 | Radio Honeywell HF 950 Installation The modification consists in installing the Bendix/King KF950 HF-1 Communications System. This includes mainly the embodiment of HF antenna, control unit, receiver, amplifier and electrical protection.
Revision 1 extends the applicability of the STC to ATR72... | STC Data List (ATR 42-300), ref: ASI-SST-0059 Issue 04, dated 04 December 2012; Aircraft Flight Manual Supplement (ATR 42-300), ref: ASI-AFM-0059, dated 07 July 2012; Instruction for Continued Airworthiness (ATR 42-300), ref: ASI-MPS-0059/Revision 01, dated October 2012; Data List (ATR 42-500), ref: ASI-S5. | This STC is limited to: MSN 148, MSN 227, MSN 559, MSN 633 and MSN 667. Prior to installation of this design change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness. | 07/06/2013 | Active | 10442513    |
<p>| 10042519    | JET AVIATION ST. LOUIS, Inc.|       | FAA STC SA02789CE   | MYSTERE FALCON MF 50        | EASA validation of FAA originally approved STC SA02789CE “Installation of a Bendix/King CAS-81 TCAS System”. The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval. See original FAA approved STC SA02789CE documents - Jet Aviation Certification Drawing List No: A919011, Revision F dated March 11th, 2011 and - Jet Aviation Report No: 13-328V001, Revision F, FAA Flight Manual Suplement dated May 13th, 2011 and - Instructions for Continued Airworthiness, Feb. 2011. | - Master Data List DC104-9041-00, Revision B, dated 22 December 2011; - Aircraft Flight Manual Supplement DC360-9041-00, Initial release, dated 19 December 2011; - EWS/ICA source document DC113-9041-00, Revision A, dated 28 October 2011; - Structural Repair Manual SRM for the B737-800 dorsal GBS radome attachment skirt as... | - Master Data List DC104-9041-00, Revision B, dated 22 December 2011; - Aircraft Flight Manual Supplement DC360-9041-00, Initial release, dated 19 December 2011; - EWS/ICA source document DC113-9041-00, Revision A, dated 28 October 2011; - Structural Repair Manual SRM for the B737-800 dorsal GBS radome attachment skirt as... | 06/12/2012 | Active | 10442519    |
| 10042528    | DELTA ENGINEERING CORPORATION | REV. 2| FAA STC ST03056CH  | 737-800                     | Validation of FAA STC ST03056CH - Installation of Panasonic Global Communications suite (GCS) System Provisions in accordance with Armstrong Aerospace Master Data List DC104-9041-00, Revision B, dated 22 December 2011 Revision 1: Transfer of STC 10038435 from Armstrong Aerospace as per FAA STC ST03056CH... | - Master Data List DC104-9041-00, Revision B, dated 22 December 2011; - Aircraft Flight Manual Supplement DC360-9041-00, Initial release, dated 19 December 2011; - EWS/ICA source document DC113-9041-00, Revision A, dated 28 October 2011; - Structural Repair Manual SRM for the B737-800 dorsal GBS radome attachment skirt as... | - Master Data List DC104-9041-00, Revision B, dated 22 December 2011; - Aircraft Flight Manual Supplement DC360-9041-00, Initial release, dated 19 December 2011; - EWS/ICA source document DC113-9041-00, Revision A, dated 28 October 2011; - Structural Repair Manual SRM for the B737-800 dorsal GBS radome attachment skirt as... | 25/01/2013 | Active | 10442528    |
| 10042532    | ELECTRONIC CABLE SPECIALISTS, Inc. |     | FAA STC ST0592CH  | B737-600-700/800-900/- 900ER | Validation of FAA STC ST02952CH - Installation of dual Electronic Flight Bag (EFB) System Provisions. The Certification Basis for the original product remains applicable to the following additional or alternative airworthiness requirements: - The associated certificate noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval. | - Master Data List (MDL), Reference ECS-200529, Revision K, FAA approved the 28 August 2012 - Instructions for Continued Airworthiness (ICA), Reference ECS-SA0357&lt;-&gt;; 26 August 2012 - EWS/ICA source document ECS-SA0357, Revision C, dated 08 July 2011 or later revisions of the above... | - Master Data List (MDL), Reference ECS-200529, Revision K, FAA approved the 28 August 2012 - Instructions for Continued Airworthiness (ICA), Reference ECS-SA0357&lt;-&gt;; 26 August 2012 - EWS/ICA source document ECS-SA0357, Revision C, dated 08 July 2011 or later revisions of the above... | 06/12/2012 | Active | 10442532    |</p>
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<td>10042553</td>
<td>STC Holder 1</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>SAS530NM</td>
<td>B737-100/200/300/400/500</td>
<td>FAA STC SAS530NM validation - Installation of the ACSS TCASIII System with Change 7.1 software</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification note/award or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>14/12/2012</td>
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<td>10042556</td>
<td>DECA AVIATION</td>
<td>REV. 3</td>
<td>FAA STC</td>
<td>ST04411SC</td>
<td>F27 MK 050</td>
<td>Installation of a Single Universal Avionics Flight Manual System with integrated GPS</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. Where amended by additional installation or removals and if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>27/03/2019</td>
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<td>10042561</td>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>ST02116AT</td>
<td>737-400, 737-500, 737-200, 737-300</td>
<td>Installation of Honeywell Ground Proximity Warning System (EGPWS) Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10042562</td>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>ST02223AT</td>
<td>737-300/400/500</td>
<td>Installation of EMTEQ Engineering dual GPS Antenna Provisions Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10042564</td>
<td>DECA AVIATION ENGINEERING Ltd.</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>SAS49-018A,</td>
<td>A320-200</td>
<td>174 Passenger - All Economy Class Passenger Seating Arrangement</td>
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<td>10042575</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>REV. 1</td>
<td>TCCA STC</td>
<td>SAS58-011BA</td>
<td>STC RC 1019</td>
<td>Installation of Left Hand and/or Right Hand Cargo Mirrors as defined in Eurocopter Canada Limited Master Drawing List MDL-95-005.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>AEROLITE AG</td>
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<td>EC835T1 (CPDS) EC135 T1, EC135 T1 (CPDS) EC135 P2, EC135 T2 EC135 P1, EC135P1 (CPDS)</td>
<td>EASA STC no. 135411-501Note: This major change to FOCASTC 25-20-45, issue 2+-+ introduces an alternative Seat/swivelling Adapter and several stowage and retainingprovisions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement FMS-204, Rev. For earlier revisions of the above listed documents approved by EASA. Master Drawing List MSL-203 Revision D Operations Manual OPM-204 Rev F (Operation and Maintenance): EASA Equipment P/N 135411-501</td>
<td>Prior to installation of this design change it must be determined whether the installation of this design change and any other previously installed design change(s) and/or repair will introduce moxhine effect upon the airworthiness of the product.</td>
<td>04/06/2014</td>
<td>Active</td>
<td>10042570</td>
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<td>AVIATION FABRICATORS, Inc.</td>
<td>REV. 3</td>
<td>FAA SA00635WI</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Beach 90 Series RH Alt Jump Seat Installation Kit Installation of RH Alt Jump Seat into the aft cargo area of thepassenger cabin. Kit includes modification to the floor structure, newfloor boards, side wall structure, and shoulder harness attachment inte aft bulkhead. Also includes additional plug ...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Installation, Operation and Maintenance in accordance with the above listed STC.</td>
<td>- Installation per STC drawing List AF-209 limited to aircraft w/Bech (Raytheon) factory installed oxygen system forjump seat installations. Aircraft configured for more than 14 total occupants must be equipped w/ 49 cubic foot (or greater) oxygen system. - Compatibility of this design change with...</td>
<td>30/06/2014</td>
<td>Active</td>
<td>10042580</td>
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<tr>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>737-800</td>
<td>Mode 5 Transponder System for Elementary and Enhanced Surveillance: 1: Transfer of STC</td>
<td>The Type and OSE Certification Bases (CB) for the original products remain applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the product unachieved and remain applicable to this certificate/approval.</td>
<td>Aviation Traders Ltd: definition Document DD/ 010/00/10 Issue 2- Aviation Traders Ltd AFM Supplement Number 23 revision 1 of the above listed document(s) approved of the EASA system.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/equipment configurations shall be determined by the installer. Prior to installation of this change/repair it must be determined that...</td>
<td>10/11/2021</td>
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<td>10042600</td>
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<td>SIERRA INDUSTRIEES Ltd</td>
<td></td>
<td></td>
<td>FAA STC SA00630AC</td>
<td>CESSNA 500</td>
<td>Installation of Meggitt Engine Indicating Display System Inaccordance with Sierra Master Drawing List, SI460-001, rev. B, dated 29 Jan 2008, Validation of FAA approval STC SA00630AC.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Sierra Master Drawing List, SI460-001, rev. B, dated 29 Jan 2008; Sierra AFM, ref. 115-601, rev. 2, dated 7 May 2007; Sierra Maintenance Manual Supplement, ref. SM-400-MMS, issue 1, dated 13 May 2004; Sierra Instruction for Continued Airworthiness, ref. SM-400- ICA, issue 1, dated 12 May 2004 or later...</td>
<td>None</td>
<td>10/12/2012</td>
<td>Active</td>
<td>10042697</td>
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<td>BERNINGER AEROT</td>
<td>REV. 2</td>
<td></td>
<td>B) DHC-6 SERIES 400 A) DHC-6 SERIES 300/30/320 A) DHC-6 SERIES 200/210 A) DHC-6 SERIES 1/101/110</td>
<td>Replacement of noise wheel BF Goodrich 8.90-12.5 PN -5-1197 by Bellmennorelle</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Modification req: DHC-6-300 STC-009 rev 04 dated 8 May 2022. Installation and Instructions for Continued Airworthiness: ICA STC-009 rev. D dated 16 Mar 2022.</td>
<td>None</td>
<td>18/03/2022</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
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<td>FAA STC STD0280AT-D</td>
<td>GV-SP</td>
<td>Installation of an Executive Completion</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable: - This certificate/approval - CRJ 0-50 &quot;reat deformation into the overwing Emergency Exit and/or Emergency Egress. - &quot; Equivalent Level of Safety-MOD-CRI F-55 'in cabin'...</td>
<td>- FAA STC STD0280AT-D issued 19th November 2012-EASA/JAR Compliance Report, ref. GC522048019, Revision - dated 15th November 2012 - Installation in accordance with Gulfstream Aerospace Corporation Index List GC522048019, Revision - dated 15th November 2012 - Operation to be in accordance with Gulfstream...</td>
<td>11/12/2012</td>
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<td>10042616</td>
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## List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>10042658</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>10042658</td>
<td>10042658</td>
<td>A321-111, A321-212, A320-211, A320-212, A319-111, A319-113,</td>
<td>Installation of the AGARS (Air Communication Addressing and Reporting System) OMU-900 to replace the Teledyne MU45B</td>
<td>Certification basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise level or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Program CP-A320-12-003 Rev. 2, dated 03/10/2012; Change Approval Sheet CAS-A320-12-003 Rev. 0, dated 19/10/2012; Airplane Flight Manual Supplement AFMS-A320-12-003-001 Rev. 0, dated 23/07/2012; Compliance Check List CCL-A320-12-003 Rev. 0, dated 11/09/2012 or later revisions of the above</td>
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<td>10042660</td>
<td>CHIPPWEWA AEROSPACE, Inc.</td>
<td>10042660</td>
<td>10042660</td>
<td>DHC-8-315, DHC-8-311, DHC-8-314, DHC-8-202, DHC-8-301, DHC-8-108, DHC-8-201, DHC-8-102, DHC-8-103,</td>
<td>Installation of a ground service bus/FAA STC ST02040AT</td>
<td>Certification basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise level or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation must be in accordance with Aviation Technologies Group Master Drawing List DHC8-001, dated 20.05.1999 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>None</td>
<td>12/12/2012</td>
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<td>10042689</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10042689</td>
<td>10042689</td>
<td>G350 AND G450</td>
<td>Installation of a Cabin Interior and Airplane Completion.</td>
<td>Certification basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval Special Condition No. 25-450-SC (Gulfstream Aerospace Corporation, Model GIV-X Airplane; Isolation of Aircraft Electronic System Security).</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List G420004225, Revision 4, dated January 01, 2012 Operation in accordance with Airplane Flight Manual Supplement G42014HM01 5, Revision 4, dated January 30, 2012, is required or later revisions of the above listed documents.</td>
<td>13/12/2012</td>
<td>Active</td>
<td>10042689</td>
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<td>10042690</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, I.L.L.C. (ACSS)</td>
<td>10042690</td>
<td>10042690</td>
<td>REV. 5</td>
<td>FAA STC ST02040LA</td>
<td>Installation of Traffic Alert and Collision Avoidance Systems (TCAS II) version 7.1. Revision to incorporate additional aircraft model to the Approved Model List 1. Hawker Beechcraft Corporation: BAE 125 Series 800xand to revise the Master Data List, to include additional interface/equipment part numbers...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce unacceptable effects upon the airworthiness of the product. The installation of this modification by third part...</td>
<td>28/11/2013</td>
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<td>10042720</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>10042720</td>
<td>10042720</td>
<td>SCT SERIES 3</td>
<td>Installation of PFD ASPEN EFD1000C3</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: Certification Program CP-A320-12-003 Rev. 2, dated 03/10/2012; Change Approval Sheet CAS-A320-12-003 Rev. 0, dated 19/10/2012; Airplane Flight Manual Supplement AFMS-A320-12-003-001 Rev. 0, dated 23/07/2012; Compliance Check List CCL-A320-12-003 Rev. 0, dated 11/09/2012 or later revisions of the above...</td>
<td>None.</td>
<td>14/12/2012</td>
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<tr>
<td>SOCATA</td>
<td>10042727</td>
<td>TB20, TB21</td>
<td></td>
<td></td>
<td>Installation of TAS Garmin GTS800</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification, noise/warn or emissions levels of the original product must not be higher and remain applicable to this certificate/approval.</td>
<td>Garmin Instructions for Continued Airworthiness for GTS 800 Traffic System, revision 2 dated 29 October 2010, socata Flight Manual Supplement S2, Edition 1 Revision 2 dated 15 November 2012, English version p/ n 200.10252201612070 socata Flight Manual Supplement S2, Edition 1 Revision 2 dated 15 November 2012.</td>
<td>This STC is not applicable to airplanes equipped with a 14 V electrical system installation. Installation of this design change must be performed by an authorized SOCATA. Personal Modified airplanes must be equipped with socata modification No. 34-001-009 approved under EASA STC No. 10029043, rev. 3 or later rev...</td>
<td>14/12/2012</td>
<td>Active</td>
<td>10042727</td>
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<td>BOND HELICOPTERS EUROPE Ltd.</td>
<td>10042728</td>
<td>EC135 T1, T2, T2+, P1, P2, P2+</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td>&quot;East Anglia Air Ambulance Role Equipment; EC 135&quot; for installation of NVIS equipment based on modification BOND/MOD/06/08 as defined in DLI/063.099.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable.</td>
<td>rotorcraft Flight Manual Supplement &quot;NIGHT VISION IMAGING/IMAGING SYSTEM ENGINE USAGE MONITORING&quot;, FMS 063-022 Issue for later revisions of the above listed documents approved by EASA.</td>
<td>Limited to Aircraft with CPFD and FODS installed. This STC applies only to the models listed above. This rotorcraft employs Night Vision Imaging System lighting, commonly named NVISighting, that has been certified to ensure the aircraft is compatible with night vision goggles (NVG). For this reason...</td>
<td>14/12/2012</td>
<td>Active</td>
<td>10042728</td>
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<tr>
<td>JCB AERO SA</td>
<td>10042732</td>
<td>A330-223</td>
<td></td>
<td></td>
<td>Galley G11 installation* Installation of Galley G11 (dry galley)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environment protection and the associated certificate noise and/or emissions levels of the original product must not be higher and remain applicable to this certificate/approval.</td>
<td>Certification Plan A332-00229-02-CP-01, Rev. 1, dated 11/December/2012 Project Data List A332-00229-02-PDL-01, Rev. B, dated 14th December/2012 Declaration of Compliance A332-00229-02-DO-01, Initial Revision, dated 12th December/2012 Instructions for Continued Airworthiness A332-00229-01-ICA-01-R...</td>
<td>The Pre-Mod status for this modification must be without seats or other furnishing.</td>
<td>14/12/2012</td>
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<td>10042732</td>
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<tr>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING ING. WALTER STARZACHER Gesa</td>
<td>10042733</td>
<td>A330-223</td>
<td>REV. 1</td>
<td></td>
<td>Cabin reconfiguration and IFE installation Description - Summary: This is a Major Change to EASA STC 10042733 to include new A/C Layout into approved configurations, and considers similar IFE System. B/C and Y/C seats. Due to different Pre-MOD cabin layout configuration the LOPA differs to the previous...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product must not be higher and remain applicable to this certificate/approval.</td>
<td>Engineering Order ED-M17300103_RSP_230 720/12, rev 1 (Mediter...</td>
<td>1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previous/installation modification and/or repair will introduce no adverse effect upon the airworthiness of the product. 2) This modification must be implemented in conj...</td>
<td>04/06/2013</td>
<td>Active</td>
<td>10042733</td>
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<tr>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>10042753</td>
<td>525, 525A, 526B 525, 525A</td>
<td></td>
<td></td>
<td>Single Installation of GPS/COM/MNAV System Garmin GTN725/750.</td>
<td>Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. CS-23, Amendment 2 (not required by AIP 15). The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product must not be higher and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement EASA-2013-053-05 Rev. 05 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previous/installation modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2012</td>
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<td>EASA STC No.</td>
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<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10042706</td>
<td>APICAL INDUSTRIES, Inc.</td>
<td>FAA STC SR02373LA</td>
<td>EC130B4</td>
<td>Installation of Fresh Air Vents in accordance with Apical Industries, Inc. Master Document List MD130-3. The STC is the validation of FAA STC SR02373LA (date of issuance: September 12, 2012).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>18/12/2012</td>
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<tr>
<td>10042768</td>
<td>HAWKER BEECHCRAFT CORPORATION</td>
<td>FAA STC SA07829W-D</td>
<td>EASA APPROVED MODEL LIST</td>
<td>Modification of Collins VHF-22A/VHF-22B VHF COMM Receiver Transmitter with CTL-22 Control Pro Line II System Installation to a Collins VHF-22C/ VHF-22D VHF COMM Receiver Transmitter with CTL-22C Control 8.33kHz Pro Line II System Installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Collins Installation Instructions/Drawings List no. AAC-0178-001 or later revisions of the above listed documents approved by EASA in accordance with FAA STC Decision 2009/04/C (or subsequent revisions of this decision)</td>
<td>A VHF-22C unit can be installed to replace either an existing VHF-22A or VHF-22B unit. A VHF-22D unit can only be installed to replace an existing VHF-22B unit.</td>
<td>18/12/2012</td>
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<tr>
<td>10042785</td>
<td>AEROSPACE DESIGN AND ENGINEERING VC NSULTANTS LTD</td>
<td>CAA UK RESPONSIBILITY</td>
<td>747-400</td>
<td>Certification of drain mast p/n 24E50789G03 against 13.1455 Boeing 747-400/ERFMSN 37299 in order to modify Hong Kong Civil Aviation Department Variation No. VAR/12/2013.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>18/12/2012</td>
<td>Active</td>
<td>10042785</td>
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<tr>
<td>10042792</td>
<td>CESSNA DUSEILBIRF CITATIONVERV E CENTER GmbH</td>
<td>S25S 525</td>
<td>WX-950 Stormscope installation.</td>
<td>CS 23 Amdt: 40The Certification Basis for the original product remains applicable to this certificate/approval Derogation to latest amendment (Part 21A, 101). Do not materially contribute to safety. The certified noise and/or emissions levels of the original product are unchanged and remain applicable.</td>
<td>- Master Data List, Doc. No. 2016-6-16, Rev. IR, dated 12-12-2012, - AFMB, Doc. No. 2016-6-10, Rev. A, dated 20-12-2012, - CAA, Doc. No. 2016-6-10, Rev. IR, dated 27-11-2012, later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>18/12/2012</td>
<td>Active</td>
<td>10042792</td>
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List of EASA Supplemental Type Certificates

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<tr>
<td>CUSTOM FIBERGLASS MOLDING INC.</td>
<td>Installation of Custom Fiberglass Molding Skis, Model 3000, according to Master Drawing List CFMI 3000-10, Rev F dated 04th April 2011.</td>
<td>No specific technical documentation referenced.</td>
<td>No specific limitations or conditions referenced.</td>
<td>Installer must check availability of EASA-approval for one or more referenced STC(s), FAA STC Approval.</td>
<td>19/12/2012</td>
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<td>10042813</td>
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<td>BEECHCRAFT BERLIN AVIATION GmbH</td>
<td>Upgrade to INCAS Mission System Suite</td>
<td>No specific technical documentation referenced.</td>
<td>No specific limitations or conditions referenced.</td>
<td>Prior to installation of this change/repair it must be determined that the installation of the above listed modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/03/2020</td>
<td>Active</td>
<td>10042830</td>
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<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>Cabin reconfiguration/By changing the existing Air Berlin Business Class Interior/Configuration into a new Elihad Business Class Interior Configuration Rev 2: Transfer of STC from SR Technics Switzerland to grOWeENGineering Partners GmbH</td>
<td>No specific technical documentation referenced.</td>
<td>No specific limitations or conditions referenced.</td>
<td>Prior to installation of this modification it must be determined that the installation of the above listed modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/04/2021</td>
<td>Active</td>
<td>10042831</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>Installation of 3-blade MT-Propeller Model MTV-0-B/198-52</td>
<td>No specific technical documentation referenced.</td>
<td>No specific limitations or conditions referenced.</td>
<td>According to limitation section of AFM-S.</td>
<td>20/12/2012</td>
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<td>STC Holder</td>
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<td>Models</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>M20R, M20S</td>
<td>Installation of 4-blade MT-Propeller Model MTV-14.D118-30b</td>
<td>The Certification Basis for the original product remains applicable tois certificate/approval. This certificate/approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/or emissions levels. Compliance with the following requirements, r...</td>
<td>According to MT-Propeller Master Document List No. E-2182 and MT-Propeller Installation Instructions Doc. No. E-2179, changes/Revisions to be operated in accordance with MT-Propeller AFMS Document No. E-2170. To be maintained i.a.w. MT-Propeller ICA Document No. E-2180, later revisions of the above li...</td>
<td>20/12/2012</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>LC41-550FG</td>
<td>Installation of 4-blade MT-Propeller Model MTV-14.D115-30b</td>
<td>The Certification Basis for the original product remains applicable tois certificate/approval. This certificate/approval involves a change to the requirements for environmental protection and/or a change to the certified noise and/or emissions levels. Compliance with the following requirements, r...</td>
<td>According to MT-Propeller Master Document List No. E-2206 and MT-Propeller Installation Instructions Doc. No. E-2203, changes/Revisions to be operated in accordance with MT-Propeller AFMS Document No. E-2201. To be maintained i.a.w. MT-Propeller ICA Document No. E-2204, later revisions of the above li...</td>
<td>20/12/2012</td>
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<td>BOMBARDIER INC.</td>
<td>STC NO C-LSA03-348/D</td>
<td>BD-700-1A10</td>
<td>Aircraft Completion, 13 Passenger Interior Configuration</td>
<td>EASA.IM.A.009: The Certification Basis for the original product remains applicable tois certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC No. C-LSA03-348/D Issue 2 dated 19 January 2004, Aeropoint Consulting Services Modification Data Summary D49100 Revision A, Bombardier Aerospace Flight Manual Supplement G- FN24000044 Rev N/C Cabin Master Power Settings G-FMD50000072 Rev A, Interior Completion G-FMD5000054 Rev N/C Smoke Detec...</td>
<td>Restricted to ASN 9093 Bombardier Service Bulletins 700-31-018 Modification - Integrated Electronics Computers - Aronics 2001 Batch 1 IAC Upgrade must be embodied.</td>
<td>20/12/2012</td>
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<td>RÖCKY MOUNTAIN AIRCRAFT</td>
<td>TCCA STC SA05-54</td>
<td>DHC-6 SERIES 200 DHC-6 SERIES 100</td>
<td>Installation of a L-3 Skywatch HP TCAS 1 System in accordance with TCCA STC SA05-54</td>
<td>The Certification Basis for the original product remains applicable tois certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement, Doc. No. E101430, TCCA approved March 19, 2012, Modification Instruction for Installation of L-3 Communications Skywatch HP Model SKY899, Doc. No. E101410, Rev. A. TCCA approved Sept. 25, 2012, later revisions of the above listed documents approved by EASA inaccordance with EASA- Modification.</td>
<td>none</td>
<td>20/12/2012</td>
</tr>
<tr>
<td>ZIM FLUGSITZ GmbH</td>
<td>REV. 1</td>
<td>B737-800</td>
<td>Installation of ZIM Flugpolz ECO-01 EC Seats in a B737-800 Cabin</td>
<td>The Certification Basis for the original product and the following additional certificated airworthiness requirements are applicable to this certificate/approval: - CRT-01 on IFD (device) 19 December 2012. The CRT-01 on IFD (device) 19 December 2012. The certified noise and/or emissions levels of the original product are unchanged and remain applicable.</td>
<td>Master Document List (MDL), Reference 55402-MDL-002, Revision 0B, dated 19 December 2012, Airplane Flight Manual Supplement (AFMS), Reference 55402-5AFM-008, Revision 0B, dated 17 December 2013, Weight and Balance Manual, Reference 55402-WBM-001, Revision 0B, dated 19 December 2012, Service Bulletin.</td>
<td>1) The In-Seat IFE system shall not be used during Taxi, Take-Off and, landing. 2) The ZIM Flugpolz ECO-01 EC Seat installation in accordance with this STC is limited to the following B737-800 aircraft: SN 41047 and 41179.</td>
<td>21/12/2012</td>
</tr>
<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>10042850</td>
<td>JET AVIATION AG, BASEL</td>
<td>REV. 1</td>
<td></td>
<td>G-SP</td>
<td>MDL DDBM002-001 Rev. IR</td>
<td>UAP-20 Change 15 and IAR AWO Change 2 with the exceptions as per EASA TCDS AM.017</td>
<td>The Certification Basis for the original product remains applicable; this certificate is approved. The certificate is noise and/or emissions levels of the original product unchanged and remain applicable to this certificate.</td>
</tr>
<tr>
<td>10042851</td>
<td>AVION RESEARCH, Inc.</td>
<td>REV. 1</td>
<td>FAA STC SA1708CH</td>
<td>2, 175, 175A/B/C, 1.72A/B/C/D/E/F/G/H/I/ K/L/M, 1.172</td>
<td>Installation of a Control Yoke and Instrument Panel in accordance with FAA STC SA1708CH</td>
<td>The Certification Basis for the original product remains applicable; this certificate is approved. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unchanged and remain applicable to this certificate.</td>
<td>None</td>
</tr>
<tr>
<td>10042854</td>
<td>AVIATION ENGINEERING AS</td>
<td>REV. 1</td>
<td></td>
<td>AS365 N1, N2, N3</td>
<td>Modification of Helidyne DPL Litter P/N HD285-15-1039 Approved under STCP/VS 2121003 (Sweden); Rebuilding of the Helidyne DPL Litter Instrument Platform P/N 007-0020633-001 (EMS Patient Platform).</td>
<td>The Certification Basis for the original product remains applicable; this certificate is approved.</td>
<td>2524-03-320 dated 25 April 2012 - Certification Plan 2524-03-322 Design Change Report</td>
</tr>
<tr>
<td>10042855</td>
<td>SCANDINAVIAN AVIONICS DESIGN AS</td>
<td>REV. 1</td>
<td></td>
<td>228-200-2011/202-212</td>
<td>Installation of GARMIN GTN750 / GTN650 and Sandel SN4500 EMS</td>
<td>The Certification Basis for the original product remains applicable; this certificate is approved.</td>
<td>- Engineering Order and Certification Program, Doc. No. 2077918-E0, Issue 1 dated 17-Dec-2012; Airplane Flight Manual Supplement, Doc. No. 2077918-AM A, Issue 2 dated 20-Dec-2012, for Garmi GTN 750 and GTN 650-Airplane Flight Manual Supplement, Doc. No. 2077918-AM B, Issue 1 dated 19-Nov-2012, for Doc...</td>
</tr>
<tr>
<td>10042873</td>
<td>CESSNA ZURICH CITATION SERVICE CENTER/GmbH</td>
<td>REV. 1</td>
<td></td>
<td>B300, B300C, B200G, B200CGT</td>
<td>BARO-NAV Temp. Comp. Activation &amp; RF Leg Authorization/Activation of Temperature Compensation on already approved APV BARO-NAV system and RF Legs Authorization (This functionality can be used in the initial and intermediate approach segments, the final phase of the missed approach; SIDs and TFAs.)</td>
<td>The Certification Basis for the original product remains applicable; this certificate is approved. The certificate is noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List: MDL-101 Rev. 2077918-EO, Suppl. B</td>
</tr>
<tr>
<td>10042880</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td></td>
<td>B200, B200C, B200G, B200CGT</td>
<td>Installation of Flight Management System Universal UniS-1W</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 2 (not required by 21A.201)(The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement ASR-2012-057-FLM-05-00 Rev. O0 dated 17-Dec-2012; for Continued Airworthiness ASR-2012-057-CA-11-01 (Rev.01) dated 17th March 2017</td>
</tr>
<tr>
<td>STC Holder</td>
<td>FAA STC STO2960-1CH</td>
<td>462C</td>
<td>CGR-P005-00 &quot;Installation of an aerial survey camera II type UltraCamX for Aerial Work Survey&quot;</td>
<td>Certification basis: The Certification Basis for the original product remains applicable with this certificate approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Foreign/NAA STC ref.</td>
<td>Description</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>COMPAGNIA GENERALA</td>
<td>FAA STC STO2946CH</td>
<td>10042960</td>
<td>CL-600 ALL MODELS</td>
<td>Report Change CGR-P-005-00 Issue 0 dated 6 August 2010, including FightManual Supplement &quot;Change CGR-P-005-00 Issue 0&quot; Rev (for later revisions of the above listed documents approved by EASA)</td>
<td>Rev.1</td>
<td>L35A, L36A</td>
<td>FAR 25 The Certification Basis for the original product remains applicable with this certificate approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
</tr>
<tr>
<td>INTERNATION &amp;</td>
<td>FAA STC STO2960-1AT</td>
<td>10042902</td>
<td>Avionics</td>
<td>ICA Instruction for continued airworthiness Document No. 5223-05Revision (for later revisions of the above listed documents approved by EASA)</td>
<td>Rev.1</td>
<td>L35A, L36A</td>
<td>Multiple Antenna Installation Provision Lear SAFTGLO, Inc. antennas are installed at the following location: 1. UHF antennas at the bottom fuselage between FR 13 and FR 17, 2. As option A: antennas located between FR 13B and 13A center fuselage.</td>
</tr>
<tr>
<td>ENGINEERINGCO</td>
<td>FAA STC STO2956-1AT</td>
<td>10042919</td>
<td>Armstrong Aerospace Inc.</td>
<td>Certification of Compatibility with Cellular and Wireless Data Technologies</td>
<td>Revision 0The aircraft shall be maintained in accordance with EASA Technical Implementation of this decision and/or the environment protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Rev.1</td>
<td>737-900ER</td>
</tr>
<tr>
<td>RPRATION AG</td>
<td>FAA STC STO2956-1AT</td>
<td>10042953</td>
<td>B/E Aerospace, Inc.</td>
<td>Installation of a Honeywell Mark VII Enhanced Ground Proximity Warning System (EGPWS) Revision 1 EMTEQ Engineering to B/E Aerospace.</td>
<td>Rev.1</td>
<td>727 SERIES</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate approval.</td>
</tr>
<tr>
<td>(AEGC)</td>
<td>FAA STC STO2995-1AT</td>
<td>10042956</td>
<td>STG Aerospace Inc AKA SAFITGOLO, Inc.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Rev.1</td>
<td>A321-212 / -232 / -231 / -230 / -220 / -221 / -211</td>
<td>Installation of SatTOG photochromic floor proximity emergency escape path marking system (FPSEPMS) - Validation of FAA STC N STIG2920AT</td>
</tr>
<tr>
<td>(AEGC)</td>
<td>FAA STC STO2995-1AT</td>
<td>10042960</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Rev.1</td>
<td>CL-600 ALL MODELS</td>
<td>Installation of Air Ambulance EquipmentRev. 1: This revision adds four additional cabin configurations to the STC and updates the STC document package to an updated Air Ambulance Technology format also considering the requirements of EASA CRD 01</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Rev.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>REV. 3</td>
<td>SA 365 N, AS 365 N2 SA 365 N1, AS 365 N3 EC 155 B, EC 155 B1</td>
<td>Installation of Air Ambulance Equipment, VIP interior, Mission Equipment, Revision 2. Introduces explicit references to the helicopter models for which this STC is applicable.</td>
<td>The Certification Basis (CB) for the original product remanufactured or modified is applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement - &quot;quick conversion&quot; PN 385-25-24.000-50P/MS 10A477 Rev. 1; - Air Ambulance Equipment &quot;quick conversion&quot; PN 385-25-24.000-51F/MS 10A477 Rev. 3; - Mission Equipment &quot;quick conversion&quot; PN. 155-25-24.000-700 model/MS 10A477 Rev. 4.; - Mio.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/06/2018</td>
</tr>
<tr>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>REV. 9</td>
<td>MBB-BK117 D-3M MBB-BK117 D-2, MBB- BK117 C-3 MBB-BK117 C-2 MBB-BK117 B-2, MBB- BK117 C-2 MBB-BK117 B-2</td>
<td>Installation of EMB interior and VIP interior called by AA-TE &quot;QuickConversion&quot; of Revision 8 and 9. To correct typographic Continuation Sheet document Revision 7. To correct the agreement applicability and document references. This revision includes conditions for the STC/applicability Revision 6.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remanufactured or modified are applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rev 6 to Rev. 9 of this STD; - Declaration of compliance DoD-D-3(m) extension of BK117 D-2(m) STC(VP) 501-Rev 0 dated 17.12.2021; - Master Data list/MDL-D3(m) extension of BK117 D-2(m) STC(VP) 501-Rev 0 dated 15.12.2021; further documents to identify applicability with regard to nosertia model/variants.</td>
<td>The EMB interior installation kit is applicable to MBB-BK117 B-2 and MBB-BK117 C-2 models. The VIP interior installation kit is applicable to MBB-BK117 C-2, MBB-BK117 D-2, MBB-BK117 D-3 and MBB- BK117 D-3m models and MBB-BK117 D-3m variant. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/01/2022</td>
</tr>
<tr>
<td>SPECTRUM AEROMEDES, LLC/ a.k.a. SPECTRUM AEROMEDES</td>
<td>REV. 2</td>
<td>FAA STC ST01612CH CL-600-2B19 CL-600-2B16 (604 VARIANT) CL-600-2B16 (601-3R VARIANT) CL-600-2B16 (601-3A VARIANT) CL-600-2A12 (601 VARIANT) CL-600-1A11 (600)</td>
<td>Air Ambulance Conversion/Reason for rev. 2: Change of ownership from Smiley Creek Enterprises, LLC to Spectrum Aeromed, LLC dba Spectrum Aeromed.</td>
<td>The Certification Basis (CB) for the original product remanufactured or modified is applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Spectrum Aeromed Inc. (+) Master Drawing List 66415L: Revision 7 dated 20 August 2012, or later FAA approved revision - Spectrum Aeromed Inc. Flight Manual Supplement 6641 PMI: Revision Taledc 18 October 2012, or later FAA approved revision - Spectrum Aeromed Inc.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/04/2019</td>
</tr>
<tr>
<td>PATS AIRCRAFT L.L.C.</td>
<td>REV. 2</td>
<td>FAA ST00936NY-0 B737-700 737-700 IGW</td>
<td>Installation of Auxiliary Fuel System *Revision 2: Added reference to MMEL Supplement 1. Configuration 1-A: Installation of structural provisions, for a Thr3 (3) Auxiliary Fuel System in the aft cargo compartment in accordance with PATS, Inc., Master Drawing List No. BBU0033, Revision 8, dated 17 February 1999 or later FAA approved revision. Configuration 1-B: Installation of strut...</td>
<td>The Certification Basis for the original product remanufactured or modified is applicable to this certificate/approval.</td>
<td>Configurations 1-A, 1-B, 1-C, 1-D, 2, 1-E, 1-F, 1-G, 1-H and 1-I. These installations are limited to five hundred airfronctypes. No liquids, gums, liquids or solids may be loaded in the tank. All configurations.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/04/2021</td>
</tr>
<tr>
<td>MTL POLYFAB Inc.</td>
<td>REV. 1</td>
<td>TCCA STC SA05-68 BD700-1A10, BD700-1A11</td>
<td>Thermal Acoustic Insulation Installation</td>
<td>The Certification Basis (CB) for the original product remanufactured or modified is applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>The aircraft must be inspected in accordance with requirement...</td>
<td>01/01/2013</td>
</tr>
<tr>
<td>STC holder</td>
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<td>Foreign/NAA Stc ref.</td>
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<td>Associated Technical Documentation</td>
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<tr>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 2</td>
<td></td>
<td>DR 400 RP</td>
<td>Installation of engine Lycoming O-540 A4DS or O-540 J3A5</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>None</td>
<td>26/01/2022</td>
</tr>
<tr>
<td>DART AEROSPACE Ltd.</td>
<td></td>
<td>TCCA STC SH 96-10 REV. 3</td>
<td>AS355 E/F/F1/F2/N/NI/P</td>
<td>Emergency Litter Installations/Installation of an Emergency Litter Kit as defined in Dart Aerospace Ltd. Master Document List MDL-D350-616.</td>
<td>The Certification Basis for the original product as amended by the following: additional or alternative airworthiness requirements FAR 27 including amendments 27-1 through 27-20, 27-807, 27-1585 and 27-1587 at amendment 27-21. The requirements for environmental protection and the associated certified noise or emissions levels.</td>
<td>Flight Manual Supplement FMS-D350-616 revision 0 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined the relationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product. The installation of this modification by third parties.</td>
</tr>
<tr>
<td>THEILERT AIRCRAFT ENGINES GmbH</td>
<td>REV. 1</td>
<td></td>
<td>MTV-6-A</td>
<td>Installation of a Constant Speed Unit CSU TAE-125 Revision 0 follows a change of ownership.</td>
<td>Manual for Installation and Operation Instruction - Constant Speed Unit CSU TAE-125.</td>
<td>MDL No. 3501-2000-1, Rev. A, dated 21/08/1998</td>
<td>1. The CSU is limited to the case when the Propeller is driven by a TAE125 engine. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft configurations shall be determined by...</td>
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<tr>
<td>TECH-TOOL PLASTICS, Inc.</td>
<td></td>
<td>FAA STC SR0900RC</td>
<td>AS355B/B/8/18/2/B3</td>
<td>Installation of miscellaneous windows in accordance with Tech-ToolPlastics, Inc. MDL 2051-2000-1. This STC is the re-issuance of LBA-Germany STC RC-1108 issued on 27 September 1999.</td>
<td>MDL No. 3501-2000-1, Rev. A, dated 21/08/1998</td>
<td>Prior to installation of this design change it must be determined the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/07/2013</td>
</tr>
<tr>
<td>TECH-TOOL PLASTICS, Inc.</td>
<td></td>
<td>FAA STC SR09003RC</td>
<td>AS 355 E/F/F1/F2/N</td>
<td>Installation of miscellaneous windows in accordance with Tech-ToolPlastics, Inc. MDL 2051-2000-1. This STC is the re-issuance of LBA-Germany STC RC-1117 issued on 27 September 1999.</td>
<td>MDL No. 3501-2000-1, Rev. A, dated 21/08/1998</td>
<td>Prior to installation of this design change it must be determined the relationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>09/07/2013</td>
</tr>
<tr>
<td>TECH-TOOL PLASTICS, Inc.</td>
<td></td>
<td>FAA STC SH98979/H</td>
<td>AS355E/F/F1/F2/N</td>
<td>Installation of aft Cabin Comfort Windows in accordance with Tech-ToolPlastics, Inc. MDL 35610C-2010-100. This STC is the re-issuance of LBA-Germany STC RC-1113 issued on 27 September 1999.</td>
<td>MDL No. 3501CC-2010-100, Rev. I.R., dated 15/07/1994</td>
<td>Prior to installation of this design change it must be determined the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/07/2013</td>
</tr>
</tbody>
</table>
This STC is approved only for installation of this change/repair on aircraft already modified by the approval holder. This STC can be installed only when in accordance with Type Design Definition ST01761NY configuration A. The installation of this change/repair must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon...
### List of EASA Supplemental Type Certificates

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<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td></td>
<td>FAA STC ST09749SC</td>
<td>GALAZY</td>
<td>Installation of a jumpseat in the forward L/H radio rack</td>
<td>- FAA STC ST09749SC - STC Data List 30-1100006, Revision C, dated April 18, 2000 or later FAA Approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft engine configuration shall be determined by the installer.</td>
<td>09/02/2005</td>
<td>Active</td>
<td>10443020</td>
<td></td>
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<tr>
<td>AIR-GLACIERS S.A.</td>
<td></td>
<td>SA 315B</td>
<td></td>
<td>Installation of external mirror This installation was previously approved by FOCA for specific SA 315B55 under FOCA Major Change MOD 03-126-011. This minor change extends the installation to SA 315B B/N 2522 and B/N 2486.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements FAR 27.210(2). The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Work Order No. 559, dated 7.10.2000Manual de vol supplement no. 20 dated 7.10.2003 later revisions of the above listed documents approved by EASA.</td>
<td>S/N: 2522, 2480 Only already existing parts manufactured and approved under FOCA MOD03-126-011 can be installed. No new part can be produced under this EASA approval. Prior to installation of this design change it must be determined through the interaction between this design change and any other previous design change.</td>
<td>23/01/2014</td>
<td>Active</td>
<td>10443023</td>
</tr>
<tr>
<td>LUTFANSA TECHNAG</td>
<td></td>
<td>757-200</td>
<td></td>
<td>Installation of a Zonal Drying System P/N BT572-02 designed and manufactured by CTT Systems (Sweden)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Madelec Aero Certification Documentation List No. CDL-9909 Issue 6 dated 19 June 2017.</td>
<td>26/01/2005</td>
<td>Active</td>
<td>10443025</td>
<td></td>
</tr>
<tr>
<td>MADELEC AERO S.A.S.</td>
<td>REV. 2</td>
<td>DGAC-F STC IM-208-SF-002</td>
<td>CL-600-2819</td>
<td>Replacement of existing Floor Proximity Emergency Exit Path Marking (FPEEM) Add new MEB in the frame of Madelec Aero STC for Air Nostrum Company Rev. 1 adds MSN 7200 to EASA STC No. 10043027 Rev. 2 states the complete applicability.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Madelec Aero Certification Documentation List No. CDL-9909 Issue 6 dated 19 June 2017.</td>
<td>Applicable to MSNs 7003, 7020, 7053, 7057, 7081, 7084, 7126, 7144, 7164, 7177, 7182, 7183, 7184, 7197, 7200, 7223, 7249, 7255, 7273, 7316, 7320, 7332, 7332, 7343, 7350, 7430, 7468, 7493, 7514, 7529, 7547, 7583, 7591, 7622, 7681, 7700, 7705, 7706, 7732, 7746, 7754, 7785, 7790, 7796, 7865, 7915, 7916, Prior to installation of this design change it must be determined through the interaction between this design change and any other previous design change.</td>
<td>06/07/2017</td>
<td>Active</td>
<td>10443027</td>
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<tr>
<td>GAMA ENGINEERING Ltd</td>
<td></td>
<td>CAA STC 820007</td>
<td>B200 AND 200</td>
<td>Installation of a PAS Incubator Part No. A0388-500-I.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>STC Definition Document: Gama Service Support Systems Ltd Substantiation Report. GAMA/SS/SR 04-001 or later revisions of the above listed documents approved by EASA.</td>
<td>28/08/2014</td>
<td>Active</td>
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An agency of the European Union
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<td>EASA R.S.00261</td>
<td>AEROLITE AG</td>
<td>REV. 7</td>
<td>EC135 T3</td>
<td>Emergency Medical Service (EMS) Cabin Configuration [initial issue as EASA R.S.00261]; Installation of EMS equipment P/N 135520-501 or 135520-502 in accordance with Master Drawing List DWL-303 dated 16 September 2004 or later/EASA-approved revision Revision 1 as EASA R.S.00261 rev.1 [Introduction of EMS equipment]</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/approval.</td>
<td>- Master Document List MDL-303 Revision F: EMS Equipment P/N 135520-501; FMS-303 Revision F: For EMS Equipment P/N 135520-501; FMS-501 Revision F: For EMS Equipment P/N 135520-501; FMS-504 Revision F: for EMS Equipment P/N 135520-501</td>
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<td>30/06/2023</td>
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<td>10043032</td>
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<tr>
<td>EASA R.S.01241</td>
<td>AEROLITE AG</td>
<td>REV. 2</td>
<td>MBB-BK 117 C-2</td>
<td>Installation of a dedicated EMS equipment P/N 145121-50-502. This STC revision is issued for the purpose of transferring the STC to AEROLITE AG.</td>
<td></td>
<td>Flight Manual Supplement FMS-506, Rev. C, dated 07.09.2006</td>
<td>Prior to installation of this design change it must be determined that the installation does not introduce a design change and/or repair and will introduce no adverse effects upon the airworthiness of the product.</td>
<td>15/03/2016</td>
<td>Active</td>
<td>10043038</td>
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<td>EASA R.S.01093</td>
<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U.</td>
<td>REV. 1</td>
<td>A109E</td>
<td>Installation of an image capture system (ARGUS 275 Camera) together with the previous installation of an SX-16 searchlight and transmission system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/approval.</td>
<td>- CMA015-AFMS Supplement for an Image Capture System (Argus275) and Transmission System - CMA015-B Engineering Bulletin - CMA015-B Installation/Maintenance Manual or later revisions of the above listed documents approved by EASA.</td>
<td></td>
<td>01/02/2019</td>
<td>Active</td>
<td>10043039</td>
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<tr>
<td>EASA R.S.01465</td>
<td>VIH AEROSPACE Inc.</td>
<td>REV. 1</td>
<td>TCMA SH08-10 ISSUE 6</td>
<td>Passenger Seating Configurations Revision 1: Reissuance of EASA STC No. EASA-R.M.S.01455 (10043046) with additional interior layouts for TCMA STC SH08-10 Issue 6.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product remain unchanged and applicable to this certificate/approval.</td>
<td>For Seating Configuration No. 1 to 18 of V92-371-PLN Revision 1: Installation drawing V92-371-PLN-FMS Revision 7 dated 29 Oct 2019. - Flight Manual Supplement for Seating Configuration No. 1 to 18 of V92-3234-PLN Revision 4 dated 22 Nov 2019. - Flight Manual Supplement for Seating Configuration No. 1 to 18 of V92-3234-PLN Revision 4 dated 22 Nov 2019.</td>
<td>Prior to installation of auxiliary fuel tank installation (approved under EASA STC 10029620) - SAR Console and operator seat (approved under EASA STC 10052472) - Triple Lifter Stacker Installation (approved under EASA STC 10053690) - GKN Sliding Door Installation (approved under EASA STC 10053690).</td>
<td>15/06/2024</td>
<td>Active</td>
<td>10043046</td>
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<tr>
<td>EASA R.S.00812</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 2</td>
<td>767-300</td>
<td>Installation of Enhanced Ground Proximity Warning system (EGPWS) Revision 1: FAR Part 26 Validation for the EASA STC EASA A.S.00612 Rev. 2. Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/approval.</td>
<td>- AES-707-146 Issue 1 - AFMS Supplement No 4 Issue 1: AES Ltd. Master Data Ltd No. AES-767-344MD, issue 5 dated 30 October 2012 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the installation does not introduce a design change and/or repair and will introduce no adverse effects upon the airworthiness of the product. This Supplemental Type Certificate is approved o...</td>
<td>27/04/2021</td>
<td>Active</td>
<td>10043054</td>
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### List of EASA Supplemental Type Certificates

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<th>EASA STC No.</th>
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<th>Foreign/NAA STC ref.</th>
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<tr>
<td>10043060</td>
<td>SPECTRUM AEROMED, LLC/ bio SPECTRUM AEROMED</td>
<td>REV. 1</td>
<td>FAA STC SA05600CH</td>
<td>PA-31T2, PA-31T3, PA-31P-350 PA-31P, PA-31T, PA-31T1</td>
<td>Air Ambulance Conversion.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Installation must be in accordance with Master Drawing List 2003DL/Rev. 3, dated 2 April 2008 or later approved revisions - Operation must be in accordance with Aircraft Flight Manual Supplement 2003FMS Rev. 1(&gt;)+1 or later approved revision - Maintenance must be in accordance with Instruction for C.</td>
<td>17/04/2019</td>
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<tr>
<td>10043062</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>REV. 2</td>
<td>FAA STC ST00670CH</td>
<td>A320-323/-341/-342/-343 A320-321/-320/-321/-322/- A320-233/243F</td>
<td>Installation of Rosemount Aerospace 8700C2-2 Laptop DockingStation - Electronic Flight Bag (EFB) System. Provision of (s):CRI H-01 &quot;Enhanced Airworthiness Programme for Aircraft - Electronic Flight Bag (EFB)&quot;. The revision 2 deals with the change of the docking station power supply source. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rosemount Aerospace Master Data List. ref : D0402872 Revision T, dated 03 March 2014 Aircraft Flight Manual Supplements - Document ref : D0402869 revision C (Configuration A before Service Bulletin 8700C2-46-01 embedmenl-Document ref : D0403259 revision C (Configuration B before Service Bulletin 8...</td>
<td>Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/11/2014</td>
<td>Active</td>
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<td>10043064</td>
<td>GDC TECHNICS, LLC.</td>
<td>REV. 1</td>
<td>FAA STC ST168115C</td>
<td>A320-214</td>
<td>Installation of an Executive Interior on Airbus A320-214MSN 3199Revision due to removal of EWIS limitation and editorial correction onprevious approval. The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- FAA STC ST100315C, amended on 29th January 2014-Gore Design Certification Drawing List. GCD-429-03, Rev. K, dated 3rd September 2006-Gore Design Substantiation Report List, GCD-429-04, Rev. C, dated 27th January 2014-Instruction for Continued Airworthiness, ref. No. GCD-429-05, Rev. C dated 25th June, 2014...</td>
<td>This STC is limited to MSN 3199 Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. The installation...</td>
<td>25/02/2014</td>
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<td>10043064</td>
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<tr>
<td>10043068</td>
<td>THUNDER BAY AVIATION Ltd.</td>
<td>REV. 1</td>
<td>TCRA SA05-30</td>
<td>DHC-6-300 DHC-6-100, DHC-6-200, DHC-6-300</td>
<td>Installation, Substitute Composite Floor Panels. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Installation to be completed as per Aero Consulting Services Modification Data Summary 2003FLM Rev. 1, dated 1 March 2008-Gore Design Certification Drawing List. GCD-429-03, Rev. K, dated 3rd September 2006-Gore Design Substantiation Report List, GCD-429-04, Rev. C, dated 27th January 2014-Instruction for Continued Airworthiness, ref. No. GCD-429...</td>
<td>Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. The installation...</td>
<td>12/05/2015</td>
<td>Active</td>
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<td>10043074</td>
<td>ACS-NAI Ltd.</td>
<td>REV. 1</td>
<td>TCRA STC C-LSAS02-020D</td>
<td>CL-600-2B19</td>
<td>Installation, Complete Custom Aircraft Interior The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Instructions for continuing airworthiness as per Aero Consulting Services Modification Data Summary 2003FLM Rev. 1, dated 1 March 2008-Gore Design Certification Drawing List. GCD-429-03, Rev. K, dated 3rd September 2006-Gore Design Substantiation Report List, GCD-429-04, Rev. C, dated 27th January 2014-Instruction for Continued Airworthiness, ref. No. GCD-429...</td>
<td>Restricted to ASN 8087.</td>
<td>16/02/2013</td>
<td>Active</td>
<td>10043074</td>
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<th>Associated Technical Documentation (EASA)</th>
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<tr>
<td>10043075</td>
<td>STC Holder</td>
<td></td>
<td></td>
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<td>Installation, Complete Custom Aircraft Interior</td>
<td>TCCA ST-175 Rev. 1</td>
<td>LS-106-007L/0-2770, Issue 2 dated 17 June 2017</td>
<td>Restricted to ASN 8004.</td>
<td>16/02/2013</td>
<td>Active</td>
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<td>10043079</td>
<td>STC Holder</td>
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<td>Installation of externally mounted survey equipment; Tail mounted survey boom &amp; optional video camera. Wing mounted pods.</td>
<td></td>
<td></td>
<td>1. Eligibility of specific items of externally mounted survey equipment must be in accordance with latest TCCA approved Eligibility List. EASA-STP-1.2. This STC approves only the structural installation of the de-ice/anti-ice system components. Separate approval is required for the activation of these systems.</td>
<td>20/06/2014</td>
<td>Active</td>
<td>10043079</td>
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<tr>
<td>10043106</td>
<td>STC Holder</td>
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<td>Extension of the existing EASA STC EASA-S-20439: Installation of MTV-14 D-C/F-P188-30/1/188-30/30. Propellor on Cessna 414A, Installation of MTV-14 D-C/F-P188-30/1/188-30/30. Propeller on Cessna 414A modified by FAA STC SAA4568SW (RAM-4) Installation of MTV-14-D-C/F-P188-30/1/188-30/30. Propeller on Cessna 414A modified by FAA STC SAA4568SW.</td>
<td></td>
<td></td>
<td>1. According to part(s) 20.02.12 of the FAA STC for Cessna Models 185, dated 10-17-83, is required for the activation of these systems.</td>
<td>07/11/2023</td>
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<td>10043115</td>
<td>STC Holder</td>
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<td></td>
<td></td>
<td>Installation of a Ground Proximity Warning System (GPWS) &amp; Terrain Awareness Warning System (TAWS). Rev. 3 transfers the STC from &quot;Deutsche Aircraft GmbH&quot; to &quot;Alamo Engineering GmbH&quot;.</td>
<td></td>
<td></td>
<td>1. Aircraft Registrations A40-CU and A40-CV. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot;.</td>
<td>15/06/2018</td>
<td>Active</td>
<td>10043115</td>
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<td>10043118</td>
<td>STC Holder</td>
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<td>Installation of a Ground Proximity Warning System (GPWS) &amp; Terrain Awareness Warning System (TAWS). Rev. 3 transfers the STC from &quot;Deutsche Aircraft GmbH&quot; to &quot;Alamo Engineering GmbH&quot;.</td>
<td></td>
<td></td>
<td>1. Aircraft Registrations A40-CU and A40-CV. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot;.</td>
<td>18/12/2023</td>
<td>Active</td>
<td>10043118</td>
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<tr>
<td>10043123</td>
<td>STC Holder</td>
<td></td>
<td></td>
<td></td>
<td>Installation of S-TEC System 40/50 single and Twin Axis Automatic Flight Guidance System, Model ST-186-40/50.</td>
<td></td>
<td></td>
<td>1. It is the installer's responsibility to ensure that the STC is approved prior to installation. This design change must be determined by the interface between this design change and any other previously installed design change or repair.</td>
<td>17/12/2013</td>
<td>Active</td>
<td>10043123</td>
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### Description

- **Certification basis**: The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.

- **Limitations/Conditions**: Revisions to the Technical Implementation Procedures of EU/Canada/Bilateral Agreement Master Drawing Lists (a) LCA0030, Rev. 5; (b) LCA00317, Rev. 3; (c) LCA00331, Rev. 1, Type Specific Panel.

- **Date of Issue**: The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.

- **Status**: It is the installer's responsibility to ensure that the STC is approved prior to installation. This design change must be determined by the interface between this design change and any other previously installed design change or repair. This STC is approved in accordance with the Amendment to the Technical Implementation Procedures of EU/Canada/Bilateral Agreement Master Drawing Lists (a) LCA0030, Rev. 5; (b) LCA00317, Rev. 3; (c) LCA00331, Rev. 1, Type Specific Panel.

### Notes

- **System Record**: 10043075, 10043079, 10043106, 10043115, 10043118, 10043123

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<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A321-212 / 213 / 231 / 232 / A321-111 / 112 / 131 / 211 / A320-211 / 212 / 233 / A320-214 / 215 / 216 / A320-111 / 112 / 113 / 131 / 132 / 133 / A319-111 / 112 / 113 / 114</td>
<td>Installation of &quot;SkyCam&quot; Video Monitoring System, Introduction of A319 type models</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amdt 4 for the paragraphs affected by the change. The requirements for environmental protection and the associated certified noise and/or emissions levels of the orig...</td>
<td>Master Data L1.J203/DOL/01, Issue 1, dated 17/12/2011 Certification Program, J203CP01, Issue 1, dated 22/11/2013</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/12/2013</td>
<td>Active</td>
<td>10043125</td>
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<tr>
<td>P&amp;V ENGINEERING</td>
<td>REV. 13</td>
<td>ATR 72-212, ATR 72-212 A, ATR 72-213 A, ATR 72-202, ATR 72-212 ATR 72-202</td>
<td>ATR72 &quot;Bulk Freighter Conversion&quot; (BFC): main cabin conversion into a cargo compartment. Rev. 2 retains the certified design of the EASA A.S.00422 and captures the STC holder change from Alenia Aermacchi to Safran Cabin Services GmbH. Rev. 3 transfers the STC from &quot;Deutsche Aircraft GmbH&quot; to &quot;P&amp;V Engineering GmbH&quot;.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: As per CRI A-01, Issue 6 dated 7 Sept 2022. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product...</td>
<td>As per AFM Supplement P2F-00090-AM3-002: Instructions for Continued Airworthiness</td>
<td>As per AFM Supplement P2F-00090-AM3-002: Instructions for Continued Airworthiness</td>
<td>21/06/2023</td>
<td>Active</td>
<td>10043128</td>
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<td>ALAMO ENGINEERING GmbH</td>
<td>REV. 3</td>
<td>JETSTREAM 4100 SERIES</td>
<td>Installation of Honeywell MK VI Enhanced Ground Proximity Warning System. STC CAA.J318E.00044. Rev. 1 transfers the STC from &quot;Deutsche Aircraft GmbH&quot; to &quot;Alamo Engineering GmbH&quot;.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remaining applicable to this certificate/ approval.</td>
<td>As per AFM Supplement P2F-00090-AM3-002: Instructions for Continued Airworthiness</td>
<td>As per AFM Supplement P2F-00090-AM3-002: Instructions for Continued Airworthiness</td>
<td>18/12/2023</td>
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<td>10043137</td>
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<tr>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 2</td>
<td>BAE.125 SERIES 800B</td>
<td>Installation of a Honeywell Airborne Collision Avoidance System (ACASI).</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remaining applicable to this certificate/ approval.</td>
<td>As per AFM Supplement P2F-00090-AM3-002: Instructions for Continued Airworthiness</td>
<td>As per AFM Supplement P2F-00090-AM3-002: Instructions for Continued Airworthiness</td>
<td>27/09/2018</td>
<td>Active</td>
<td>10043139</td>
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<tr>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 2</td>
<td>DORNIER 328-100</td>
<td>Installation of a Honeywell MK VI Enhanced Ground Proximity Warning System (EGPWS).</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remaining applicable to this certificate/ approval.</td>
<td>STC Definition Document</td>
<td>STC Definition Document</td>
<td>27/09/2018</td>
<td>Active</td>
<td>10043140</td>
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<td>SAFRAN CABIN INC.</td>
<td>REV. 1</td>
<td>FAA STC ST01406A</td>
<td>Installation of a reinforced cockpit door.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remaining applicable to this certificate/ approval.</td>
<td>C &amp; D Interiors Controlled Drawing List, 7152000, Revision E, dated 29/November 2002, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. Equipment Functional Check intervals as defined...</td>
<td>06/04/2018</td>
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<td>EASA STC No.</td>
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<td>Description</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10043150</td>
<td>TELEDYNE CONTROLS, LLC</td>
<td>REV. 3</td>
<td>FAA STC ST01620LA</td>
<td>A321-211N, A320-251N, A220-321N, A321-211H, A320-251N, A220-321H, A321-211J, A320-251N, A220-321J, A321-211L, A320-251N, A220-321L</td>
<td>Installation of Teledyne Controls Wireless Ground Link Quick Access Recorder. This STC supersedes EASA IM.A.S.0092</td>
<td>Master Data List MDL2243800-32X, revision 2, dated 8 March 2004</td>
<td>Prior to installation of this design change it must be determined whether the installation of this design change and any other previously installed design change or repair will introduce new or increased noise or emissions or increase noise or emissions levels of the original product. The STC becomes invalid if submitted noise or emissions levels of the original product exceed the noise or emissions levels of the original product as called out in the respective Master Drawing List MDL2243800-330 Appendices.</td>
<td>01/10/2019</td>
<td>Active</td>
<td>10043150</td>
</tr>
<tr>
<td>10043151</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 1</td>
<td>ST01632LA</td>
<td>757-200</td>
<td>Aerospace Engineering Solutions Modification AES-757-165 Issue 6 Installation of Enhanced Ground Proximity Warning System (EGPWS) Revision 1: Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>Master Data List Drawing No. MDL2243800-757, revision 2, dated 3 March 2004, or later FAA approved revisions.</td>
<td>This STC is approved only for the product configuration defined in the approval data referenced to in the paragraph &quot;Description.&quot; Compatibility with other aircraft or configurations shall be determined by the installer. This STC shall remain valid unless otherwise surrendered.</td>
<td>28/04/2021</td>
<td>Active</td>
<td>10043151</td>
</tr>
<tr>
<td>10043159</td>
<td>TELEDYNE CONTROLS, LLC</td>
<td>REV. 2</td>
<td>ST01632LA</td>
<td>757-200, 758-200F, 757-300</td>
<td>Installation of Teledyne Controls Wireless Ground Link Quick Access Recorder.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined whether the installation of this design change and any other previously installed design change or repair will introduce new or increased noise or emissions or increase noise or emissions levels of the original product.</td>
<td>12/04/2016</td>
<td>Active</td>
<td>10043159</td>
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<tr>
<td>10043160</td>
<td>TELEDYNE CONTROLS, LLC</td>
<td>REV. 4</td>
<td>FAA STC ST01632LA</td>
<td>A330-343, A330-302, A330-301</td>
<td>Installation of Teledyne Controls Wireless Ground Link Quick Access Recorder.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined whether the installation of this design change and any other previously installed design change or repair will introduce new or increased noise or emissions or increase noise or emissions levels of the original product.</td>
<td>11/09/2018</td>
<td>Active</td>
<td>10043160</td>
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<tr>
<td>10043181</td>
<td>TELEDYNE CONTROLS, LLC</td>
<td>REV. 4</td>
<td>FAA STC SAR030NM</td>
<td>737-800/-900ER, 737-700/-800, 737-100/200/300/400</td>
<td>Requirement for the installation of Teledyne Controls Digital Flight Data Acquisition Unit (DFDAU) and Digital Flight Data Management Unit (DFDMU). This STC supersedes UK CAA's certificate CAA.21ME.0003T.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined whether the installation of this design change and any other previously installed design change or repair will introduce new or increased noise or emissions or increase noise or emissions levels of the original product.</td>
<td>07/10/2019</td>
<td>Active</td>
<td>10043181</td>
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<td>STC Holder</td>
<td>STC No</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Information</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>TELEDYNE CONTROLS, LLC</td>
<td>ST0264LA</td>
<td>737-800</td>
<td>Installation of Teledyne Controls Wireless Ground Link Quick Access Recorder (WGL-QAR). This STC superseded Luftfahrt-Bundesamt (LBA) certificate LBA.2151.E.UA743 and follows EASA associated STC letter 2004-0796 see in change in applicants address.</td>
<td>The Certification Basis for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged remain applicable to this certificate / approval.</td>
<td>Teledyne Controls MDL 2243800-737, Revision &quot;B&quot;, Appendix A, Nov. 16, 2001 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/04/C (or subsequent revisions off this decision) and/or the Technical Implementation Procedures of EULASA Bilateral Agre...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aeroplane.</td>
<td>12/04/2016</td>
<td>Active</td>
<td>10103162</td>
<td></td>
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<tr>
<td>TELEDYNE CONTROLS, LLC</td>
<td>F4S152</td>
<td>MD-88 DC-9-37 (MD-87) DC-9-83 (MD-83) DC-9-82 (MD-82) DC-9-81 (MD-81)</td>
<td>Validation of STC Teledyne Controls Aircraft Quick Access Recorder (OQAR) P/N 2248060-40 on McDonnell Douglas MD-80. This STC was previously approved by Registro Aeronautico Italiano.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged remain applicable to this certificate / approval.</td>
<td>FAA approved Teledyne Controls Master Data List No. MDL 2248060-40, Revision &quot;B&quot;, dated November 10, 1993 later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/04/C (or subsequent revisions off this decision) and/or the Technical Implementation Procedures of EULASA Bilateral Agre...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aeroplane.</td>
<td>12/04/2016</td>
<td>Active</td>
<td>10103163</td>
<td></td>
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<tr>
<td>ATELIER CHABORD</td>
<td>F152, FA152</td>
<td>152</td>
<td>Exhaust system and silencer installation (according to Chabord Modification STC CESSNA 152-17-01) Revision 1: Changes in raw material and coating. Revision 2: Enhanced alarm block.</td>
<td>The Certification Basis for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged remain applicable to this certificate / approval.</td>
<td>Technical Dossier ref. STC CESSNA 152-17-01 Revision B dated 09 June 2002 for later revisions of the above listed document(s) approved / accepted by the EASA system.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 21.A.105. Prior to installation of this change/repair, the interrelationship of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aeroplane.</td>
<td>27/07/2022</td>
<td>Active</td>
<td>10103169</td>
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<tr>
<td>ACS-NAI Ltd.</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Installation, Aircraft Communication System</td>
<td>EASA IM.A226The Certification Basis for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged remain applicable to this certificate / approval.</td>
<td>FAA STC SAR-0171</td>
<td>Restricted to ASN 5541</td>
<td>09/01/2013</td>
<td>Active</td>
<td>10103177</td>
<td></td>
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<tr>
<td>EUROCOPTER UK Ltd.</td>
<td>EC 135 T+</td>
<td>Installation of RTE Aircraft Installation of LH and RH Meeker steps. EASA STC EASA (M.R.S. S1408). ArSWE Systems Camera system is installed with Turret attached to the front of the LH Hand Meeker Step via Meeker A11 Utility Mount. Meeker GT-1 Dovetail adaptor and EC UK Adapter Pa...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged remain applicable to this certificate / approval.</td>
<td>- Transport Canada STC No. C-LASA12-046D, Issue A dated 12 July 2012.-ACS-NAI Ltd Modification Data Summary H048000 Revision NC.-ACS-NAI Ltd Mandatory Maintenance Manual Supplement for the Installation, Indium Antenna FS 580, HX40TS Revision NC.-ACS-NAI Ltd Instructions for Continued Airworthiness...</td>
<td>- Design Report N° DR 159407-01, Issue A, dated 7 December 2012.- Instructors 159407-01, Issue A, dated 14 December 2012.- Supplement N° 60 to the approved EC 135 T+ VFR. Installation RTE\Ecopt, Issue A, dated 18th December 2012.</td>
<td>For Day/Night VFR only.</td>
<td>09/01/2013</td>
<td>Active</td>
<td>10103181</td>
<td></td>
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<tr>
<td>PRATT &amp; WHITNEY ENGINE SERVICES, IN</td>
<td>F4S152</td>
<td>DHC-8-401/402 DHC-8-311/311-314/315 DHC-8-201-202</td>
<td>Installation of a P &amp; W Flight Data Acquisition, Storage and Transmission System (FAST) extension to models DHC-8-200 Series</td>
<td>The Certification Basis for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged remain applicable to this certificate / approval.</td>
<td>Installation in accordance with Installation Data List of GN4-1.100-1, Rev 8 or GN4-1.100-2, Rev 8. Operation in accordance with AFMS GN4-1.250-1, Rev. A or GN4-1.250-3 Rev. A. Instructions for Continued Airworthiness in accordance with GN4- T-260-2, Rev. B or GN4- T-260-3, Rev. C, or later revisions of the...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aeroplane.</td>
<td>20/10/2014</td>
<td>Active</td>
<td>10104007</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10043211</td>
<td>SABENA TECHNICS BOD</td>
<td></td>
<td></td>
<td>A320-214</td>
<td>Installation of new 9101VU &amp; 9102VU racks in cargo area</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- Certification Plan PC-DD087-003 Issue 3 dated 07/01/2013-Master Drawing List MDL-S-53-2269-01 Issue A dated 03/01/2013-Instructions for Continued Airworthiness (ICA) ICA-DD087-00-003 Issue A dated 07/12/2012-Modifications Approval SheetMAS-S-53-2269 Issue 4 dated 07/01/2013</td>
<td>This approval is limited only to MSN 764</td>
<td>11/02/2013</td>
<td>Active</td>
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<tr>
<td>10043212</td>
<td>SABENA TECHNICS BOD</td>
<td>REV. 1</td>
<td></td>
<td>A320-214</td>
<td>Structural provisions for Live TV Antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- Certification Plan PC-DD087-001: Issue 7, dated 16/09/2013-Master Drawing List MDL-S-53-2270, Issue D, dated 03/01/2013-Instructions for Continued Airworthiness (ICA) ICA-DD087-00-003, Issue B dated 05/09/2013-Airworthiness Limitation Supplement ALS-S-53-53-001, Issue B, dated 02/07/2015-Modifications...</td>
<td>This approval is limited to MSN 794 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/09/2013</td>
<td>Active</td>
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<tr>
<td>10043216</td>
<td>PRO STAR AVIATION, LLC</td>
<td></td>
<td>FAA STC SA0032180</td>
<td>PC-12/46, PC-12/47, PC-12/47E</td>
<td>STC Prostar Aviator 200 inst. on PC-12/46/47/47E for Communication with WiFi and Phone through SATCOM.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate approval. FAA MoC PT S-1 stage 4 &quot;Installation WLAN&quot;. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>FAA AFMS &quot;Aviator 200 SwiftBroadband Sys&quot; Doc. 21037-FMS-01Rev. If other later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>S/N: all legacy MSN</td>
<td>14/01/2013</td>
<td>Active</td>
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<tr>
<td>10043218</td>
<td>SABENA TECHNICS BOD</td>
<td></td>
<td></td>
<td>A320-214</td>
<td>Installation of new forward INMARSAT SATCOM system, all SATCOM SBUpgrade, new Live TV system and upgrade of the IFE and CMS</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- Certification Plan PC-DD087-002 Issue 3 dated 07/01/2013-Master Drawing List MDL-S-53-2250 Issue A dated 07/01/2013-Instructions for Continued Airworthiness (ICA) ICA-DD087-00-001 Issue A dated 06/12/2012-Modifications Approval SheetMAS-S-53-2230 Issue 4 dated 07/01/2013-Modifications...</td>
<td>This approval is limited only to MSN 794 This modification can be installed only concurrently with the modifications approved under EASA STC 10043212 (EASA project 0010016696).</td>
<td>11/01/2013</td>
<td>Active</td>
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<tr>
<td>10043232</td>
<td>INNOTECH AVIATION/A DIVISION OF IMP GROUP LTD</td>
<td></td>
<td>TCCA STC SA12-63</td>
<td>BD-700-1A10</td>
<td>Installation, Aircraft Completion Modifications.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25Amendment 1 - Various. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged on...</td>
<td>Transport Canada STC No. SA12-63, Issue 1, dated 17 October 2012-Innotech Aviation Configuration Definition Report CER-BD700-001 RevisionA-Innotech Aviation-Airplane Flight Manual/Supplement AFMS-IA6000-2500-AI2-Instructions for Continued Airworthiness Document GC9620000-ICA or later revision...</td>
<td>Restricted to ASN 9452 - The WLAN System is intended to provide internet connectivity and e-mail services to the airplane's cabin using Portable Electronic Devices (PEDS). Any other intended function of this equipment will require a re-examination of the certification basis.</td>
<td>14/01/2013</td>
<td>Active</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10043246</td>
<td>AERO-DENST GmbH &amp; Co. KG</td>
<td>REV. 1</td>
<td>10043264</td>
<td>700-10-160 (CHALLENGER 300)</td>
<td>Installation of a Three-Bladed MT-Propeller Model</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Engineering Order 284-CL.056.01 rev 1 dated 29/12/2012, Instructions for Continued Airworthiness ref. AG959-085 rev IR, Airplane Flight Manual Supplement AFMS AG959-071 rev IR</td>
<td>Refer to AFMS AG959-071 rev IR.</td>
<td>16/01/2013</td>
<td>Active</td>
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<tr>
<td>10043261</td>
<td>FIELD AVIATION EAST, LTD.</td>
<td></td>
<td>TCDC STC D-LSA12-130/D</td>
<td>DHC-8-202</td>
<td>Dual Universal VNS-1(E)(x)-Flight Management System Upgrade</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation must be performed in accordance with FAA Flight Manual Supplement No. 25673000 Rev. 1(1)&lt;br&gt;Operations must be performed in accordance with FACI Flight Manual Supplement No. 25673000 Rev. 1(1)&lt;br&gt;Maintenance must be performed in accordance with FACI Maintenance Manuals.</td>
<td>Limited to Serial Numbers 477, 488, 496, 498 and 504 with EASA STC No. 10028595 (Advanced Avionics Technology) installed.</td>
<td>16/01/2013</td>
<td>Active</td>
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<tr>
<td>10043262</td>
<td>NSE INDUSTRIES S.A.</td>
<td></td>
<td></td>
<td>A109E</td>
<td>Installation of Night Image Vision System kit, allowing night VFR flight with Night Vision Goggles</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- AEROSRACT FLIGHT MANUAL SUPPLEMENT A109 - document RFMS-0020-12AFLP Rev 04r later revisions of the above listed documents approved by EASA - CONFORMITY STATEMENT AND DOCUMENTARY ARCHITECTURE A109, Document Reference CONFAD-101A-1000.000 Rev 01-INSTRUCTION FOR CONTINUED AIRWORTHINESS NVIS A109E; do.</td>
<td>Refer to applicable RFM for list of equipment mandatory for flight operations with NVIS/ NVG compatible lighting does not include or imply approval for flight operations with Night Vision Goggles (NVG). The operator must receive approval for the Civil Aviation operations.</td>
<td>17/01/2013</td>
<td>Active</td>
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<tr>
<td>10043263</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td></td>
<td></td>
<td>G 115E</td>
<td>Installation of 3-blade MT-Propeller Model MTV-12-B/C/C183-17e</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection and a change to the associated certified noise and/or emissions levels of the original product as well as Compliance with the following requirements:</td>
<td>According to MT-Propeller Master Document List No. E-2266 and MT-Propeller Installation Instructions Doc. No. E-2264, Revisions To be performed in accordance with MT-Propeller AFA-5 Document No. E-2260 To be maintained i.a.w. MT-Propeller ICA Document No. E-2250 and later revisions of the above listed documents.</td>
<td>according to limitation section of AFA-5</td>
<td>17/01/2013</td>
<td>Active</td>
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<tr>
<td>10043265</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>FAA STC NO. ST02198CH</td>
<td>717-200</td>
<td>Installation of Rosemount Aerospace Inc. 8400E1-Cockpit Door Surveillance System in Boeing 717-200 Series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA STC NO. ST02198CH dated 28 October 2005, Rosemount Aerospace Master Data List, Doc. No D5503319, Revision B, dated 28 October 2005, FAAMS 8400E Flight Manual Supplement, Rosemount Aerospace Doc. No. D550100, Revision A, dated 28 October 2005 or later revisions of the above listed documents approved by.</td>
<td>none</td>
<td>17/01/2013</td>
<td>Active</td>
<td>10043265</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>STC Holders</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>10043267</td>
<td>FAA STD- STO0504NY</td>
<td>REV. 2</td>
<td>777-200-200LR/300-300</td>
<td>Installation of a Panasonic Avionics Corporation Global Communications Suite (GCS)Rev. 1 that adds the 777-200-200LR to the list of applicable models</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: For the 777-200-200LR models. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following...</td>
<td>- Master Data List 0143-0764-2282 Rev. Q dated 19 Nov 2016. - FAA Supplement 0643-07850-2282 Rev. C dated 20 Oct 2016. - FAA Instructions for Continued Airworthiness.</td>
<td>FAA STC (EASA Approval) as validated by EASA STC Ref. 10043256-A.</td>
<td>13/02/2019</td>
<td>Active</td>
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<tr>
<td>JET AVIATION AG, BASEL</td>
<td>10043269</td>
<td></td>
<td></td>
<td>MF50 (EX-VERSION)</td>
<td>WLAN activation Electro-Magnetic Compatibility demonstration of airplane’s requirement systems immunity to worst case WLAN (IEEE 802.11 a/b/g/n) passengercarry-on flight WLAN transmitters.</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The certificated noticeable or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List Doc. No. BFSLA001-02 Rev. IR dated 11.01.2013 - AFMS. Doc. No. BFSLA001-02-02 Rev. IR dated 11.01.2013 - ICA. Doc. No. BFSLA001-150-02 Rev. IR dated 10.12.2012 later revisions of the above listed documents approved by EASA.</td>
<td>FAA STC (EASA Approval) as validated by EASA STC Ref. 10043256-A.</td>
<td>17/01/2013</td>
<td>Active</td>
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<tr>
<td>AVIONIK STRAUBING ENTWICKLINGS GmbH</td>
<td>10043270</td>
<td>MBB-BK 117 B-1, B-2, C-1, C-2, MBB-BK 117 A-1, A-3, A-4</td>
<td>Installation of Cockpit Voice Recorder L-3 FZ2100CVR as defined in ASR-2012-093-PRD-04-00.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>ASR-2012-093-TC-P06-00 dated 06/11/2012/ASR-2012-093-PRD-04-00 dated 14/01/2012/ASR-2012-093-COR-07-00 dated 10/01/2013/ASR-2012-093-MDL-01-00 dated 10/01/2013.</td>
<td>None</td>
<td>FAA STC (EASA Approval) as validated by EASA STC Ref. 10043256-A.</td>
<td>17/01/2013</td>
<td>Active</td>
<td>10043271</td>
</tr>
<tr>
<td>AVIONITEC Ltd.</td>
<td>10043286</td>
<td>MBB-BK 117 B-1, B-2, C-1, C-2, MBB-BK 117 A-1, A-3, A-4</td>
<td>Installation of Cockpit Voice Recorder L-3 FZ2100CVR as defined in ASR-2012-093-PRD-04-00.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>ASR-2012-093-TC-P06-00 dated 06/11/2012/ASR-2012-093-PRD-04-00 dated 14/01/2012/ASR-2012-093-COR-07-00 dated 10/01/2013/ASR-2012-093-MDL-01-00 dated 10/01/2013.</td>
<td>None</td>
<td>FAA STC (EASA Approval) as validated by EASA STC Ref. 10043256-A.</td>
<td>17/01/2013</td>
<td>Active</td>
<td>10043289</td>
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<tr>
<td>A Y M AVIONICS, SL</td>
<td>10043289</td>
<td>PA-34-200</td>
<td>Installation of Dual COMNAV/GPS GARMIN GNS 430 &amp; GNS 530</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Technical Documents to be approved by EASA as SERVICE BULLETIN N°: 19/07 R1 - SMV N° 19/07 R1 - Airplane Flight Manual Supplement for IFR operation with Dual Garmin GNS 430W installation.</td>
<td>FAA STC (EASA Approval) as validated by EASA STC Ref. 10043256-A.</td>
<td>19/02/2013</td>
<td>Active</td>
<td>10043289</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>KAMAN CORPORATIONAL RCRRAFT WHEEL AND BRAKE, LLC</td>
<td>2- A200C, A200CT, B200CT, B200T, 2-200, 200C, 200CT, 2007,A200 1- B200, B200C, B200GT,B200GT 1</td>
<td>FAA STD- SAB906L</td>
<td>Installation of Parker Hannifin Corporation, Aircraft Wheel &amp; Brake A/Eurocopter Division Hot Air Manifold Kx 199-110, revision AB, dated June 18, 2010, and applicable, installation of Parker Hannifin Corporation, Aircraft Wheel &amp; Brake A/Eurocopter Division Hot Air Manifold Kx 199-110, revision F, dated June 18, 2010.</td>
<td>The Certification Basis for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained applicable to this certificate/approval.</td>
<td>Installation in accordance with Installation Drawing 50-79, revision N, dated June 18, 2010, Continued Airworthiness in accordance with Parker Hannifin Corporation, Component Maintenance Manual CM30-146, revision D, dated January 19, 2011, and Component Maintenance Manual CM40-269, revision B, dated Aug., 2012.</td>
<td>This installation is restricted to those aircraft with standard landing gear (18x5.5 main Wheel &amp; Brakes).</td>
<td>18/01/2013</td>
<td>Active</td>
<td>10043296</td>
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<tr>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>A330-202</td>
<td>Stretcher installation and Aircraft cabin reconfiguration.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25: Amendment 4 - Various: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged after the revision.</td>
<td>EASA STC 10043300H4 Aerospace Modification Summary and Declaration of Compliance No.: MOD44A1967 Issue 4 dated 10 March 2013/4 Aircraft Flight Manual Supplement No.: H4AFM0249H4 Instructions for Continued Airworthiness No.: H4ADR7560k later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interruption between this design change and any otherwise previously installed design change and/or repair will introduce nonadverse effect upon the airworthiness of the product. The installation of this modification by third party.</td>
<td>10/04/2013</td>
<td>Active</td>
<td>10043300</td>
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<tr>
<td>EUROCOPTER CANADA Ltd.</td>
<td></td>
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<td>EC120B</td>
<td>Door Handle installation (P/N 120-200014/Validation of TCCASTC SH12-0 Issue 1 dated 09/02/2012) to correct handling problems induced by the serial design of the right-hand cabin door handle. Eurocopter Canada Limited has designed a new door handle.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>Definition and Installation in accordance with:</td>
<td>None</td>
<td>18/01/2013</td>
<td>Active</td>
<td>10043307</td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC STO3878AT</td>
<td>757-300 757-200</td>
<td>Installation of the Spectralux DLink + ACARS System into Boeing 757-200/300 Series Aircraft and associated hardware. Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Drawing List 2-S9475-001 Revision A, dated January 22, 2002. FAA Approved Airplane Flight Manual Supplement for the SpectraluxDLink+ ACARS, 2-S9475-805, revision IR, dated March 11, 2009. Instructions for Continued Airworthiness (CA) EMTEQ Engineering/Document No. 2-S9475-902, Rev. A, da...</td>
<td>Prior to installation of this design change it must be determined that the interruption between this design change and any otherwise previously installed design change and/or repair will introduce nonadverse effect upon the airworthiness of the product.</td>
<td>02/03/2016</td>
<td>Active</td>
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<tr>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 1</td>
<td>CESSNA 340A CESSNA 340</td>
<td>Alternative Supplemental Inspection Document (A-SID) Rev. 1 Transfer to ACC Columbia Jet Service GmbH</td>
<td>- According to Gomolzig MDL Document No. MDOL210100_REV0_300 42009 printed/EASA approved Revisions- To be operated and maintained in accordance with approved Gomolzig Document No. 3428_A-SID_Master Manual20100_rev0 or later EASA approved Revisions 1. Alteration to A-SID is subject to written permission of approvalholder2. This modification is only limited to SIN listed in EASA approved Gomolzig Document No.20100_Master Aircraft Application Control3. Ltd. 30042009, rev0, or later EASA approved Revisions. Prior to installation of this modification...</td>
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<td>26/01/2022</td>
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<td>10043338</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STD SAJ971AT</td>
<td>DHC-6 SERIES 1, 190, 200, 300</td>
<td>DHC-6</td>
<td>Replace NiCd or Lead Acid battery with Concorde Lead Acid battery and associated hardware in accordance with Master Drawing List 5-0398 Revision E, dated May 4, 2012 or later FAA approved drawing revisions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Instructions for Continued Airworthiness (ICA), Concorde Battery Corporation No. S-0446, Rev. NC, dated September 29, 2011 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Im...</td>
<td>none</td>
<td>22/01/2013</td>
<td>Active</td>
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<td>10043346</td>
<td>BERINGER AERO</td>
<td>REV. 7</td>
<td>SR20, SR22, SR22T</td>
<td></td>
<td>Installation of main wheels (RF-006(A)), nose wheel (RA-003(A)) and brake system (EA-003.4N(A)) Revision 1: Upgrade of the brake system: EA-003.4N (B) superseded EA-003.4N(A) Revision 2: Update of maintenance manual and part list revision 3: Maintenance manuals updates and introduction of an aircraft flight.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>AFM Supplement: AFMS-STC-004 rev 01 dated 21 Feb. 2022; Installation and Instructions for Continued Airworthiness: ICA-STC-004-05 dated 19 Oct. 2021; Modification file: DB-STC-004 rev 08 dated 11 Mar. 2022 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/04/2024</td>
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<td>10043347</td>
<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A310-304</td>
<td></td>
<td>Avionics and Cabin Systems Upgrade</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-01, Glass Screens of Displays/Monitor CRH-01, Enhanced Airworthiness Programme for Airlplane Systems – rCOn EWIS: Specified paragraphs from FAR 25 Amendment 54 (see...</td>
<td>Modification TEN 687, Airbus A310-304 Avionics and Cabin Systems Upgrade, Issue 2 dated 17 January 2013; Report TEN 6871, Certification Compliance Report, Avionics and Cabin Systems Upgrade, Issue 2 dated 17 January 2013; or later revisions of the above listed documents approved by EASA.</td>
<td>Aircraft Serial Number 431 only.</td>
<td>22/01/2013</td>
<td>Active</td>
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<td>10043355</td>
<td>WESTERN AVIONICS</td>
<td>TCCA STC SH11-20 ISSUE 1</td>
<td>R44 &amp; R44 II</td>
<td></td>
<td>Installation of Aspen Avionics EFD 1000H/1000H Electronic Flight Display Systems in accordance with Western Avionics Engineering Order CE13-401.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: FAR 27.810 at Amend. 27; The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and rem...</td>
<td>Flight Manual Supplements: Configuration, ... Supplements, ... A approval DateEFD1000H PFD, ... CE13-801, ... 20 11 June 2013 EFD1000H PFD &amp; EFD1000H MFD, CE13-802, ... 2011 June 23 EFD1000H PFD &amp; EFD1000H MFD, CE13-803, ... 2011 June 23, or later revisions of the above listed documents approved by EASA.</td>
<td>none</td>
<td>23/01/2013</td>
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<tr>
<td>10043560</td>
<td>DAIR T AEROSPACE Ltd.</td>
<td>REV. 3</td>
<td>TCCA SH12-58 ISSUE 5</td>
<td>429</td>
<td>INSTALLATION OF EXTERNAL ATTACHMENT PROVISIONS: CARGO BASKET AND CABINSTEP CONFIGURATION A - External Attachment Provisions Only: Installation afedefined in DCL959-2, CONFIGURATION B - External Cargo Basket: Installation as defined in DCL959-1 and modifications eligible in accordance with DCL704-429.CO...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>CONFIGURATION A: Aero Design Ltd. Document Control List, DCL959-2, Rev. 2, dated 12 September 2014; CONFIGURATION B: Aero Design Ltd. Document Control List, DCL959-1, Rev. 2, dated 12 September 2014; (if), (r), (y) Aero Design Ltd. Document Control List, DCL704-429, Rev. 1, dated 12 September 2014; CONFIGURATIONS A...</td>
<td>1. Installation of Configuration A, external attachment provisions, is a prerequisite for installation of Configuration B, external cargo basket. 2. Installation of Configuration A, external attachment provisions, is a prerequisite for installation of Configuration C, cabin steps, 3. Installation of th...</td>
<td>10/11/2021</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10043374</td>
<td>JCB AEROSA</td>
<td>1</td>
<td>1</td>
<td>2E78-600</td>
<td>VIP Cabin Reconfiguration Project Data List B738-35792-01-PDL-01 rev G</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the specimens as listed in the data list.</td>
<td>Aircraft Flight Manual Supplement B738-35792-01-AFM6-01 New Instructions for Continued Airworthiness - B738-35792-01-AOA1-01 rev C Cabin Attendant Operating Manual B738-35792-01-AOM0-01 rev C</td>
<td>24/01/2013</td>
<td>Active</td>
<td>10043374</td>
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<tr>
<td>10043375</td>
<td>BOMBARDIER INC.</td>
<td>1</td>
<td>1</td>
<td>BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9381</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements CRI D-680 Shower Installation</td>
<td>TCCA STC No. C- LSA12-011 Issue 1 dated 17 February 2013 Bombardier Inc. Modification Data Summary RAZ-M6701-001-MDS Revision NC Airplane Flight Manual Supplement for the Aircraft Completion C- FM2560001 Revision NC Instructions for Continued Airworthiness for the Aircraft Interior Completion, C919</td>
<td>24/01/2013</td>
<td>Active</td>
<td>10043375</td>
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<tr>
<td>10043377</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>2</td>
<td>1</td>
<td>777-300</td>
<td>Interior Reconfiguration for Cathay Pacific Boeing 777-300 Aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and noise emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List Interior Reconfiguration Cathay Pacific Airways Boeing 777-300 Aircraft - NOA5-D10 Revision B Airways Boeing 777-300 Aircraft Instructions for Continued Airworthiness - Principle Manual for an Interior Reconfiguration on Cathay Pacific AirWay, LTD. (CAT) B777-300 Aircraft - N.</td>
<td>01/02/2013</td>
<td>Active</td>
<td>10043377</td>
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<td>10043384</td>
<td>SPECTRUM AEROMED, LLC/ ABA SPECTRUM AEROME</td>
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<td>1</td>
<td>PC-12, PC-1245, PC-1247</td>
<td>Air ambulance conversion in accordance FAA STC SA02488CH.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and noise emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Flight Manual Supplement, Spectrum Aeromed Manual 604S, Initial Release, dated September 21, 2007 Instructions for Continued Airworthiness, Spectrum Aeromed Manual 804MM, Initial Release, dated May 30, 2007 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/04/2019</td>
<td>Active</td>
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<tr>
<td>10043385</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>2</td>
<td>1</td>
<td>AB139, AW139</td>
<td>Replacement of NICAD battery with Concorde Lead-Acid battery. Note: This STC revision is done for transfer of original EASA STC 10043385 Rev 1 to new holder, Concorde Battery Corporation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and noise emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>For Rotorcraft with RG-38E4 XL main battery installed DECA Aviation Engineering Flight Manual Supplement FM311143-01 revision 2 dated July 10, 2012 Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>If the holder agrees to permit another person to use this certificate holder as a product, the holder must give the other person written evidence of that permission Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously...</td>
<td>22/03/2016</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<td>STC Holder</td>
<td>REV. 1</td>
<td>206B, L, L-1, L-3, L-4</td>
<td>Installation of Usage Monitoring System Brite Saver 2, recorder box P/N REC-ARA0-T1-00 up to Amdt 07.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unrelated and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement Bull. 206, RFA5-ARA0-REC-206-5EH +revision 00 or later revisions of the above listed documents approved by EASA Instruction for Continued Airworthiness ICA-REC-ARA0-206 revision 00 (noALB associated with this STC) Mounting Instruction M-KIT-ARA0-206-EN revision 090open...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>04/02/2016</td>
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<td>10043392</td>
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<tr>
<td>LUFTHANSA TECHNIS AG</td>
<td>REV. 1</td>
<td>A319-112</td>
<td>Installation of NEK cabin modification.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unrelated and remain applicable to this certificate/approval.</td>
<td>Declaration of Compliance ADC 3499 Rev. IR, dated 18/01/2013...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>25/01/2013</td>
<td>Active</td>
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<tr>
<td>SABENA TECHNICS BBO</td>
<td>REV. 2</td>
<td>ATR 42-600</td>
<td>Cabin and Communication systems upgradeThis modification consists in installing SATCOM and cabin telephony/capacity, as well as a modification of the rear galleys area (new galleysand seat arrangement configuration). The Revision 1 relaxed the initial terms of limitations, from 3 to 6months. The Flavia...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unrelated and remain applicable to this certificate/approval.</td>
<td>Modification Approval Sheet, ref : MAS-S-00-2282</td>
<td>The embodiment of this STC is limited to MSN 859 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>24/07/2013</td>
<td>Active</td>
<td>10043408</td>
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<tr>
<td>E.A.D AEROSPACE AIRWORTHINESS</td>
<td>REV. 1</td>
<td>A340-541</td>
<td>Description for the initial approval Installation of GSM System in an A340/GSMConnex is solution that provides GSM services to aircraft passengersand crew. It consists of hardware and software parts for both the airway and the ground, withthe airborne GSM network being connected to the ground network via...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unrelated and remain applicable to this certificate/approval.</td>
<td>Documentation related to the initial approval not totally superseded bythe Aircraft Flight Manual Supplement, ref : AFMS-S4014-00-011 Issue 3, dated 17 January 2013/Instruction for Continued Airworthiness, ref : ICA-01-C. As is stated in this document the certification basis for the new product remains unchanged and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>This approval is limited to the aircraft MSN 465 (A340-541) configurated according to the ICA document 0138-EAS-A-ICA-01-C. As is stated in the above mentioned documents the aircraft must be equipped only with those elements provided to be compatible with the GSMConnex installed.For Cockpit and Cabin CIDS handsets, the P...</td>
<td>06/05/2013</td>
<td>Active</td>
<td>10043418</td>
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<tr>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES A.R.L</td>
<td>REV. 1</td>
<td>C90A, C90GT, C90GTi, B90, ES90, CS90</td>
<td>STC 3807/12 revision 3 - Installation STC EASA 10037574</td>
<td>The Certification Basis for the original product and the followingadditional or alternative airworthiness requirements are applicable to this STC/approval: CS 23 Amdt 2/7/8 and/ or relevant sections of the initial STC.</td>
<td><a href="http://easa.europa.eu">EASA.europa.eu</a></td>
<td>The STC is approved for aircraft equipped with EN/ENI and automaticips defined in GARMIN IM 190-0107-A3 and 190-0107-I-02 and STC 2807/12annex C2 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installe...</td>
<td>28/01/2013</td>
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<td>Model</td>
<td>Certification</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>ARMSTRONG AEROSPACE Inc.</td>
<td>FAA STC</td>
<td>ST0348CH</td>
<td>B777-200-200LR/300-300/300-300</td>
<td>Certification of Compatibility with Catia\r\r\r and Wireless Data Technologies</td>
<td>STC Twenty One Limited Project No. 2104-0010-04-00, Revision E dated 21 December 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA Ed. 200404/00/0 (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU USA via EASA STC Twenty One Limited Project No. ST02257SE</td>
<td>- Rockwell Collins TCAS 4000 installation of True North Simphone connection with a separate design change. The design change introduces a satellite antenna to the crown skin of the aircraft. The design change is limited to the mechanical changes to the aircraft only. The antenna will be used in connection with a separate design change. The antenna will be used in connection with a separate design change. Prior to installation of this modification, it must be determined that the aircraft total load electrical consumption does not exceed 80% of the total aircraft electrical capacity. Installation of new control panels.</td>
<td>This modification is compatible only with main rotor blades equipped with tracking fingers in 13448-20 or 355A11-2013-20. The lay-down rules are pleased to indicate that they are not to be used in winds above 40 knots. Prior to installation of this installation change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft total load electrical consumption does not exceed 80% of the total aircraft electrical capacity. Installation of new control panels.</td>
<td>18/04/2016</td>
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<td>10043426</td>
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<tr>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>REV. 2</td>
<td>TCAC</td>
<td>EC 130 B4, EC 130 T2, AS 350 N, AS 355 NP, AS 355 E, AS 355 F, AS 355 F1, AS 350 B, AS 350 B1, AS 350 B2, AS 350 B3, AS 350 BA, AS 350 D</td>
<td>Certification of Compatibility with Catia\r\r\r and Wireless Data Technologies</td>
<td>STC Twenty One Limited Project No. 2104-0010-04-00, Revision E dated 21 December 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA Ed. 200404/00/0 (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU USA via EASA STC Twenty One Limited Project No. ST02257SE</td>
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<td>This modification is compatible only with main rotor blades equipped with tracking fingers in 13448-20 or 355A11-2013-20. The lay-down rules are pleased to indicate that they are not to be used in winds above 40 knots. Prior to installation of this installation change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft total load electrical consumption does not exceed 80% of the total aircraft electrical capacity. Installation of new control panels.</td>
<td>18/04/2016</td>
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<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>B200, B200C, B200T, B200, B200T, BCA 139 Revision B dated 28 Jan. 2013</td>
<td>Certification of Compatibility with Catia\r\r\r and Wireless Data Technologies</td>
<td>STC Twenty One Limited Project No. 2104-0010-04-00, Revision E dated 21 December 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA Ed. 200404/00/0 (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU USA via EASA STC Twenty One Limited Project No. ST02257SE</td>
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<td>This modification is compatible only with main rotor blades equipped with tracking fingers in 13448-20 or 355A11-2013-20. The lay-down rules are pleased to indicate that they are not to be used in winds above 40 knots. Prior to installation of this installation change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft total load electrical consumption does not exceed 80% of the total aircraft electrical capacity. Installation of new control panels.</td>
<td>30/01/2013</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>FAA STC</td>
<td>ST0333CH-D</td>
<td>G-115A, G-115, G-115, G-115, G-115, G-115</td>
<td>Certification of Compatibility with Catia\r\r\r and Wireless Data Technologies</td>
<td>STC Twenty One Limited Project No. 2104-0010-04-00, Revision E dated 21 December 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA Ed. 200404/00/0 (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU USA via EASA STC Twenty One Limited Project No. ST02257SE</td>
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<td>This modification is compatible only with main rotor blades equipped with tracking fingers in 13448-20 or 355A11-2013-20. The lay-down rules are pleased to indicate that they are not to be used in winds above 40 knots. Prior to installation of this installation change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft total load electrical consumption does not exceed 80% of the total aircraft electrical capacity. Installation of new control panels.</td>
<td>30/01/2013</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC</td>
<td>ST02257SE</td>
<td>777-200</td>
<td>Certification of Compatibility with Catia\r\r\r and Wireless Data Technologies</td>
<td>STC Twenty One Limited Project No. 2104-0010-04-00, Revision E dated 21 December 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA Ed. 200404/00/0 (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU USA via EASA STC Twenty One Limited Project No. ST02257SE</td>
<td>- Rockwell Collins TCAS 4000 installation of True North Simphone connection with a separate design change. The design change introduces a satellite antenna to the crown skin of the aircraft. The design change is limited to the mechanical changes to the aircraft only. The antenna will be used in connection with a separate design change. Prior to installation of this modification, it must be determined that the aircraft total load electrical consumption does not exceed 80% of the total aircraft electrical capacity. Installation of new control panels.</td>
<td>This modification is compatible only with main rotor blades equipped with tracking fingers in 13448-20 or 355A11-2013-20. The lay-down rules are pleased to indicate that they are not to be used in winds above 40 knots. Prior to installation of this installation change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft total load electrical consumption does not exceed 80% of the total aircraft electrical capacity. Installation of new control panels.</td>
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<td>01/04/2014</td>
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<td>1004372</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA AIR RESPONSIBIL ITY</td>
<td>DHC-8-401, DHC-8-402</td>
<td>Introduction of CabinVu 123 Cabin Surveillance System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation must be in accordance with STC Twenty One Master Document, S211-02-16, Issue 5. Maintenance must be in accordance with Supplement Manual 211-TEC-321, Issue 3. Operation must be in accordance with AD Aerospace Operating Manual CabinVu 123, latest issue or later revisions of the above.</td>
<td>None</td>
<td>31/01/2013</td>
<td>Active</td>
<td>1004372</td>
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<td>1004377</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>A330-323-341-342-343 A330-301-302-303-321-322 A330-201-202-203-223-243</td>
<td>A330 Air China WiFi System</td>
<td>The modification deals with the installation of a WiFi system on Air China A330 aircraft. The WiFi system is built with 2 LRUs installed Electronic Bay (AMEP) and PnE) and 7 WiFi Access Points dispatched in the passage cabin area. The Power Over Ethernet (PoE) hosts an Ethernet...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List AE102/FMS-3, Rev. 0, dated 22/01/2013; Flight Manual Supplement AE102/FMS-1, Rev. 0, dated 22/01/2013. Instructions for Continuing Airworthiness AE102/CA Rev. 0, dated 30/01/2013 or later revisions of the above documentation approved by EASA.</td>
<td>None</td>
<td>31/01/2013</td>
<td>Active</td>
<td>1004377</td>
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<td>1004381</td>
<td>VIDEO TECHNOLOGY SERVICES, Inc.</td>
<td>FAA ST03223NY</td>
<td>B767-200/-300</td>
<td>Installation of LCD Monitors.</td>
<td>The Certification Basis for the original product as amended by the systems requiring additional or alternative airworthiness requirements: 25.305, 25.307, 25.561, 25.603, 25.605, 25.613, 25.625, 25.785, 25.789, 25.853, 25.1391, 25.1529 at Amendment 11; The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Technical Drawing List 2005- D-480, Revision - 4, dated 1/02/2013; ICA Document 03/2012-ICD Document 205-202-481, Revision B, dated 15/07/2012 or later revisions of the above documentation approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Proc.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved only.</td>
<td>31/01/2013</td>
<td>Active</td>
<td>1004381</td>
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<tr>
<td>10043502</td>
<td>DASSAULT FALCON SERVICE</td>
<td>F2000, F200EX</td>
<td>LIFEPORT Powered Loading System Installation/Patient Lift System Installation/Patient Lift System Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>DFS Change Approval Sheet DFS153-CAS Issue A dated 31.01.2013; DFS Modification Sheet CFS153, Issue A dated 25.01.2013 or later revisions of the above documentation approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved only.</td>
<td>01/02/2013</td>
<td>Active</td>
<td>10043502</td>
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<tr>
<td>10043505</td>
<td>BUSINESS AND COMPUTER AIRCRAFT</td>
<td>REV. 1</td>
<td>2, B200, B200C, B200T, B200GT</td>
<td>BCA 136 Revision 0 dated 28 Jan 2013 - Flight Display Systems FDO200CPU-Band FD0152C Moving Map cabin installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement BCA136-02 Rev 0 dated 28 Jan 2013 or later revisions of the above documentation approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved only.</td>
<td>02/03/2015</td>
<td>Active</td>
<td>10043505</td>
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<td>10043526</td>
<td>STANDARD AERO</td>
<td>FAA STC ST01175CH-D</td>
<td>A330-342, A330-343 A330-322, A330-323, A330-341, A330-301, A330-302, A330-303, Installation of the ARINC Cabin Connect system on-board Airbus A330-300 series aircraft. The connection or activation of the components installed is not covered by this Design Change. It must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved only.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List DC/104-9157-00, dated 01/07/2013; FAA Approval ST03175 CH-D, dated 30/01/2013; Statement of Compliance, dated 03/10/2013; Foreign Valid 3175 EASA received 01/02/2013 or later revisions of the above documentation approved by EASA in accordance with EASA ED Decision 2004/04/CF (or sub...)</td>
<td>The connection or activation of the components installed by this modification is not covered by this Supplemental Certificate. Prior to installation of this modification it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved only.</td>
<td>04/02/2013</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>BELL 429</td>
<td>Bell 429 Human External Cargo Hook modification Dual Cargo Hook Described in Master Data List</td>
<td>- Flight Manual Supplement No.1 aeroDAC Modification Number/JSB2 Dual Cargo Hook, Issue 1, dated 04 February 2013 or later revisions of the above listed documents approved by EASA. - Master Data List Dual Cargo Hook JNSB2 Issue 1, 10 January 2013. - Installation Instruction Dual Cargo JNSB2 Issue 2.</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the: interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved o...</td>
<td>04/02/2013</td>
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<td>10043534</td>
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<td>MBB-BK117 C-2</td>
<td>Replacement of the Flir analogue video recorder SONY DSR-20MDP by digital video recorder SKYQUEST VORi-3000-01, as detailed in M004-85-11-145.</td>
<td>- Dossier de Modification M004-85-11-145 Rev. 2. - Installation instructions and Instructions for Continued Airworthiness: Service Bulletin: &quot;Installation VRDV EC149&quot;, 145HU-SB-65-69 Rev. 1, or later revisions of the above listed documents.</td>
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<td>Prior to installation of this design change it must be determined that the: interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved o...</td>
<td>04/02/2013</td>
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<td>10043536</td>
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<td>AS 355 N, AS 355 NP, AS 355 F1, AS 355 F2, AS 355 E, AS 355 F, AS 350 B3, AS 350 B1, AS 350 B2,</td>
<td>Installation of an airborne high-pressure spray system (Simplex Model510) capable of cleaning debris from the</td>
<td>- Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>14/02/2013</td>
<td>Active</td>
<td>10043539</td>
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<td>Installation of an airborne high-pressure spray system (Simplex Model510) capable of cleaning debris from the insulators on electrical power transmission towers. The system consists mainly of an external water tank, a power pack and a tumult新一代喷雾系统。</td>
<td>- Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>(M004-85-11-145).</td>
<td></td>
<td></td>
<td>02/03/2016</td>
<td>Active</td>
<td>10043542</td>
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<td>F/A STC 0398AT</td>
<td>Installation of the Honeywell MCS-7147 SATCOM System/DESCRIPTION: Removes and replaces the existing Honeywell MCS-600 arsCS-700 SATCOM System’s Satellite Data Unit (SDU) and HighPower Amplifier (HPA) with the MCS-7147 SATCOM System’s SDU and HPA/High Speed Data Terminal (HSTD). Revision 2 trans...</td>
<td>- Master Drawing List 2-95373-001 Revision B, dated September 12, 2012. - Instructions for Continued Airworthiness (ICA). - EMTEQ Engineering/Designing No. 2-95373-804. Revision A, dated September 12, 2012or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 200414CF or (subsequent revisions of this decision) and/or the Technical Implementation Procedures of EUAAS Bilateral Airworthiness MLD10-001, Rev. H or later revisions/P5.</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the: interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved o...</td>
<td>12/02/2013</td>
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<td>10043552</td>
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<td>AS 380, B300C (KING AIR) B200, B200C, B200GT, B200GT</td>
<td>Installation of a radar and a turret on a B380ER.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>Master Document List: PP-026-MDL-001Instructions for Continued Airworthiness: PP-026-ICA/01Flight Manual Supplement: PP-026-ICM-970 later revisions of the above listed documents approved by EASA.</td>
<td>Limited to S/N FL7083. Limitations according to Flight Manual Supplement PP-026-ICM-970. Prior to installation of this design change it must be determined that the: interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved o...</td>
<td>10/06/2015</td>
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<td>F/A STC 0005N9</td>
<td>Installation of a Panasonic Ku-Band Radome / Antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. ST0205N9 Data Engineering Master Data List No:0143-0735-2274 Rev. 1 dated 12 February 2014.</td>
<td>None</td>
<td>06/10/2015</td>
<td>Active</td>
<td>10043559</td>
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List of EASA Supplemental Type Certificates

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<tr>
<td>STC 1043560</td>
<td>GAMA AVIATION (ENGINEERING) LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>206L-4, 206L-4, 407 206A, 206B, 206L-1</td>
<td>Split bilateral Environmental Protection (SDEP) system for Bell206/407 Installation 1281, ID in MDL-1281, and installation 1387, ID in MDL-1387, are based upon the Thales Sweep 20 infra-red sensor suite. This revision also extends the applicability of these installations to the B...</td>
<td>The Certification Basis for the original product as amended by incorporating additional or alternative airworthiness requirements Installation 1387. EASA CS-27 amendment 3 (as in Certification ProgramCP-1281 Issue 2) Installation 1387. EASA CS-27 amendment 3 (as in Certification ProgramCP-1387 Issue 1...</td>
<td>For installation 1381: Rotorcraft Flight Manual Supplement Reference/RFMS-1281-01 Issue 0/2 Installation 1387 Rotorcraft Flight Manual Supplement Reference/RFMS-1387-01 Issue 0/1or later revisions of the above listed documents approved by EASA For installation 1281 Master Document List MDL-1281 issue...</td>
<td>20/05/2015</td>
<td>Active</td>
<td>1043560</td>
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<tr>
<td>STC 1043568</td>
<td>JAMCO AMERICA, INC.</td>
<td>FAA STC ST01494SE</td>
<td>777-200, 777-200LR</td>
<td>Lavatory and Galley Modification</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>EASA STC no. JAMCO America Inc. Master Drawing List A-11MDL, Revision C, dated December 29, 2004 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions off this decision) and/or the Technical Implementation Procedures of EUUSA B...</td>
<td>07/02/2013</td>
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<td>STC 1043585</td>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>DH-8-401, DH-8-402</td>
<td>Dual NAVDAR EFB class 2 System Provisions</td>
<td>Dual NAVDAR EFB class 2 System Provisions</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation must be in accordance with Technical Order Doc. 2068119-TO Issue 3 Operation must be in accordance with AFM Supplement Doc. 2068119-AM A Issue 3 or Doc. 2068119-AM B ISSUE 3 Maintenance must be in accordance with MM Supplement Doc. 2068119-MM ISSUE 3 or later revisions of the above...</td>
<td>This STC does not constitute a Technical Approval. Individual Technical Approvals, including the COTS equipment suitabilty, must be issued by the competent NAA. Prior to installation of this design change it must be tested that the interrelationship between this design change and any other previously tested design...</td>
<td>08/02/2013</td>
<td>Active</td>
<td>1043585</td>
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<tr>
<td>STC 1043587</td>
<td>PMV ENGINEERING</td>
<td>A330-341, A330-342, A330-343, A330-321, A330-322, A330-323, A330-301, A330-302, A330-303, A330-221, A330-231, A330-243, A330-201, A330-202, A330-203</td>
<td>Installation of ECS EZMount mechanical mounting on Airbus A330</td>
<td>Installation of ECS EZMount mechanical mounting on Airbus A330</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Certification Investigation Report AVI/00250-CIR, Revision 0, dated 04/09/2012 and Report AVI/00250-CIR, Revision 0, dated 04/09/2012 for later revisions of the above listed documents approved by...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously tested design...</td>
<td>08/02/2013</td>
<td>Active</td>
<td>1043587</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>FEDERATION FRANÇAISE DE VOL A VOILE (FFVV)</td>
<td>Polyurethane paint - M02-12-A01</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise/emissions level of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental to Manual d'Entretien C101 PEGASE Rev 1.1.doc and later revision(s) of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved o...</td>
<td>12/02/2013</td>
<td>Active</td>
<td>1043603</td>
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<tr>
<td>FEDERATION FRANÇAISE DE VOL ENPLANEUR</td>
<td>Polyurethane paint - M03-12-A01</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise/emissions level of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental to Manual d'Entretien ASW 20 F FL Rev 1.2 for later revision(s) of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This Supplemental Type Certificate is approved o...</td>
<td>12/02/2013</td>
<td>Active</td>
<td>1043604</td>
<td></td>
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<tr>
<td>KUERZI-AVIONICS AG</td>
<td>AutoNav upgrade with Garmin S500/removed equipment.</td>
<td>The Certification Basis for the original product and the following voluntary compliance to FAR 27 was shown. The actual Amdt. used for each § is indicated after the FAR §. e.g. FAR 27, 1317 / 42 means Amdt 42 was used.</td>
<td>Flight Manual Supplement MNS-200932756, Original issue, dated 11.02.2013 for later revision(s) of the above listed documents approved by EASA.</td>
<td>S/N: all - At least one SBAS capable Garmin GPS and must be installed as per requirement. This STC can only be installed in a helicopter with a left hand command PP - Instrument Panel (no slim line panel). Prior to installation of this STC the airworthiness of the product is determined by the STC holder.</td>
<td>14/02/2013</td>
<td>Active</td>
<td>1043605</td>
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<tr>
<td>CORPORATION LOGIC AIR Inc.</td>
<td>Installation of Dual CMC SBAS Landing System with CMA-5024 GLSSU sensors and Dual CMC CMA-9265 control panels. Configuration 1: Dual CMC GPS sensor (RNP) system. Configuration 2: Dual CMC SBAS landing system (SL5) Rev 1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/applicable to this certificate/approval without any limitations.</td>
<td>Configuration 1: Installation Document List MDL-34436-2 rev 24 Aircraft Flight Manual Supplement MMS-34436-1 rev 2 and Maintenance Manual Supplement MMS-34436-1 rev 2.Aircraft Flight Manual Supplement MMS-34436-1 rev 2 and Maintenance Manual Supplement MMS-34436-1 rev 2</td>
<td>Configuration 1: Limitations as per AFMS 34436-1 Configuration 2: Limitations as per AFMS 34436-1 and adapter NAV-550 P/N 550-0000, installed by drawing 0-34280-5, must be at modification status 2 (or later revision(s)). Prior to installation of this change/repair it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/06/2021</td>
<td>Active</td>
<td>1043626</td>
<td></td>
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<tr>
<td>ANLVU</td>
<td>Implementation of provisions of EFB Class 2 installation of provisions for detachable Electronic Flight Bag Class 2 on both pilot and co-pilot stations. Rev. 2: Transfer of STC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged/applicable to this certificate/approval.</td>
<td>- Master Data List AM004-MDL, Rev. 3, dated 08/02/2013 Aircraft Flight Manual Supplement AM004-AM, Rev. 1. dated 01/11/2012 - Instructions for Continued Airworthiness AM004-CA, Rev. 0, dated 02/08/2012 - Maintenance Manual Supplement AM004-IM, Rev. 1, dated 07/11/2013 for later revision(s) of the above listed documents</td>
<td>The display, the software and the applications included in the device installed are not covered by this approval. This Supplemental Type Certificate is approved only for the product/configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; The installation of this m...</td>
<td>05/06/2021</td>
<td>Active</td>
<td>1043627</td>
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<td>STC Holder</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC Ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AVIATION ENGINEERING AS</td>
<td>B300</td>
<td>REV. 2</td>
<td></td>
<td></td>
<td>Installation of FIS UNIFIS 3000 Flight Inspection System Transfer from Beechcraft Berlin Aviation GmbH to Aviation Engineering</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain unaltered or unchanged.</td>
<td>Master Document:UNIFIS Doc. No. D11-B17-09, Issue 02, dated 18-Dec-2019;</td>
<td>For limitations see AFMS. Doc. No. D30-B17-09. Prior to installation of the change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/02/2023</td>
<td>Active</td>
<td>10043630</td>
</tr>
<tr>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>B300</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>Alternate Position for ACAS and XPDRI Antennas</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Engineering Order: 265-BE. 056.01 Rev. 2, or later approved revisions. Instructions for Continued Airworthiness: AG62-085 IR. Or later approved revisions. Declaration of Compliance for Project 265-BE 08.02.2013.</td>
<td>Restricted to S/N FL-101. FL-130. LBA STC SA 1358 must have been already installed. Prior to installation of this designchange it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness.</td>
<td>21/02/2013</td>
<td>Active</td>
<td>10043637</td>
</tr>
<tr>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>O-540-J3A</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>Use of Automotive Gasoline RON 98 per EN228 for Lycoming Engine O-540-J3A5 (S/N L24488-40A) Rev. 1. STC Transfer from Gomolzig Maschinenbau GmbH to ACC Columbia Jet Services GmbH</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment - FAR 33 Amendment 18F for the following affected areas: 33.1 Applicability, 33.3 General, 33.5 Instruction manual...</td>
<td>Instructions for operation, maintenance and inspection: Gomolzig Supplement to Lycoming Operator's and Maintenance Manual SOM020738, initial issue or later revisions of the above listed document(s) approved by the EASA system.</td>
<td>The data in the FAA TCDS E-265 remains valid, except the following: Fuels Aviation Gasoline, 100, 100LL or 100VLL per ASTM D910, Automotive Gasoline RON 98 per EN228 (max 1% alcohol), see Gomolzig Supplement to Lycoming Operator's and Maintenance Manual SOM 202738, initial issue or later approved revisions...</td>
<td>17/01/2022</td>
<td>Active</td>
<td>10043650</td>
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<tr>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>A1096</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>Installation of Max-Vu EVS-1500 infrared camera.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The modification is compliant with the requirements of JAR 27 Amendment 3 paragraphs: 25, 29, 65, 67, 145, 171, 173, 177, 179, 201, 301, 303, 305, 307, 341, 361-691, 603, 653...</td>
<td>- RFM Supplement doc. 25105F101554-0601 Revision 0, dated 15.01.2013 or later revisions of the above listed documents approved by EASA. Master Document List doc. 25105F101554-0109 Revision 3, dated 15.01.2013 - Instructions for Continued Airworthiness etc.</td>
<td>None.</td>
<td>25/02/2013</td>
<td>Active</td>
<td>10043684</td>
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<tr>
<td>NATIONAL AIRCRAFT SERVICES, INC.</td>
<td>747-400F</td>
<td>REV. 3</td>
<td>747-200C, 747-300, 747-400, 747-500, 747-200B, 747-200F</td>
<td></td>
<td>NASI Pack Flow Control System Modification of the Pneumatic Air Cycle Systems by a National AircraftServices, Inc B747 Flow Control System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered or unchanged.</td>
<td>Master Data List, NKC-242; Revision NC-Aircraft Flight Manual Supplement, NKC-2424AMS, Revision DICA; Maintenance Manual Supplement, NK-747, NK-52, Revision AEWIS Compliance Plan NK-803; Repair 18 Revision A101, later revisions of the above listed documents approved by EASA in accordance with EASA AD Decision 200...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third parties...</td>
<td>18/02/2013</td>
<td>Active</td>
<td>10043676</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10043705</td>
<td>BRITISH AIRWAYS P/BRITISH AIRWAYS ENGINEERING/CORP. OF AIRWORTHINESS</td>
<td></td>
<td></td>
<td>B767-300</td>
<td>Installation of new TCAS components on all B767-300 aircraft</td>
<td>Certification Basis for the original product as amended by subsequent TCAS changes and/or associated airworthiness requirements</td>
<td>Master Data List BA-767-34-0009-MDL</td>
<td>prior to installation of this design change must be determined. Interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/02/2013</td>
<td>Active</td>
<td>10043705</td>
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<tr>
<td>10043713</td>
<td>ANCRA INTERNATIONAL LLC</td>
<td></td>
<td></td>
<td>MD-11F</td>
<td>Installation of mechanical cargo loading system Main Deck on all MD-11F aircraft</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated airworthiness requirements of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Ancra International LLC Controlled Data List DL81990 Rev 2 (dated 9/21/2011) and FAA approved Ancra International LLC</td>
<td>prior to installation of this design change must be determined. Interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/02/2013</td>
<td>Active</td>
<td>10043713</td>
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<tr>
<td>10043722</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 2</td>
<td></td>
<td>767-300</td>
<td>Installation of new IFE System Rev 2 on all DOA AES Global 767-300 aircraft</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated airworthiness requirements of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AES Ltd. Master Data List No. AES-767-2525 MDL Issue 8</td>
<td>prior to installation of this design change must be determined. Interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/04/2021</td>
<td>Active</td>
<td>10043722</td>
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<tr>
<td>10043724</td>
<td>AIRCRAFT STRUCTURES INTERNATIONAL CORP.</td>
<td>REV. 2</td>
<td></td>
<td>CESSNA 208/208B</td>
<td>Installation of a Standby Flap Travel Limiting System in accordance with FAA STC SA11115SC on all Cessna 208/208B aircraft</td>
<td>Certification Basis for the original product as amended by subsequent TCAS changes and/or associated airworthiness requirements / subordinating paragraph(s) at a later amendment: FAR 23.305(a)(b), Amdt. 49; FAR 23.1319(d)(ix), Amdt. 49. The requirements for environmental protection and the associated airworthiness requirements of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List AS1208-6000-01MDL, Revision 1, released September 22, 2011</td>
<td>prior to installation of this design change must be determined. Interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/02/2013</td>
<td>Active</td>
<td>10043724</td>
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<tr>
<td>10043733</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td></td>
<td>GVI (G650)</td>
<td>Installation of a Broadband Multi-Link System on Gulfstream GVI aircraft</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated airworthiness requirements of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- FAA STC ST0425AT-D - GDL 1000000044(-P) Revision P, dated 10 November 2021; - FCC00022635, Revision -; EASA Validated Configurations List ST0425AT-D - Airplane Flight Manual Supplement GC61309M001, Revision B, dated 16 January 2013; - FAA STC ST0425AT-D - Airplane Flight Manual Supplement GC61303A001, Revision B, dated 10 February 2013.</td>
<td>prior to installation of this design change/repair it must be determined. Interrelationship between this design change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/03/2022</td>
<td>Active</td>
<td>10043733</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
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<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10043740</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC</td>
<td>ST01493SE</td>
<td>777-200-200LR</td>
<td>Interior Reconfiguration: Installation of business class seats (&gt;7) &gt; new RH &amp; LH Nicols, fold down tables, side walls up, wash and ceiling downwash, full spectrum, digital LED lighting system, new economy class quintuple seats economy class double seats and all supporting interior and structural...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Drawings Document Flight Structures Document ASJFS237-D15, Revision A, dated January 6, 2005 - Aircraft FlightManual Supplement, Flight Structures Document TP2-237AFM, dated...</td>
<td>EASA validated version of FAA STC ST01493SE for installation of elevated and STC ST020440NY for installation of an inflight entertainment system must be installed concurrently with this STC. Prior to installation of this design change, it must be determined that this...</td>
<td>21/02/2013</td>
<td>Active</td>
<td>10043740</td>
</tr>
<tr>
<td>10043741</td>
<td>ASI INNOVATION</td>
<td>B300</td>
<td></td>
<td></td>
<td>Replacement of the Universal FMS UNS-1M by the FMS UNS-1E (including P-RNAV capability).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Service Bulletin, ASI Innovation document ASI-STB-B350-3500-0106, Revision 01, dated February 13, 2013 - Aircraft FlightManual Supplement, ASI Innovation document ASI-APM-0106, Revision 0, dated October 2012 - Instructions for Continued Airworthiness Supplement, ASI Innovation...</td>
<td>Limited to Serial Number FL.129. Prior to installation of this design change, it must be determined that the...</td>
<td>21/02/2013</td>
<td>Active</td>
<td>10043741</td>
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<tr>
<td>10043743</td>
<td>AIRLIFT A/S</td>
<td>AS 350 B, BA, B1, B2, B3</td>
<td></td>
<td></td>
<td>Installation of REMOVABLE EXTERNAL MOUNT described in AIRLIFT A/S Design Order DO-350-25-115 Rev. 0 dated 01 June 2011.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>AIRLIFT A/S Rotorcraft Flight Manual Supplement FMS-350-25-115-01 Rev. 5 dated 20 February 2013 or later revisions of the above listed documents approved by EASA Design Order DO-350-25-115 Rev. 0 dated 01 June 2011 - Instructions for continued Airworthiness ICA-350-25-115-01 Rev. 3 dated 20 February 2013...</td>
<td>OPERATIONS WITH A LOAD ATTACHED TO THE REMOVABLE MOUNT MUST BE CONDUCTED IN ACCORDANCE WITH THE APPROPRIATE OPERATING RULES FOR EXTERNAL LOADS. Prior to installation of this design change, it must be determined that the...</td>
<td>21/02/2013</td>
<td>Active</td>
<td>10043743</td>
</tr>
<tr>
<td>10043755</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC</td>
<td>ST04277AT-D</td>
<td>GVI (G650)</td>
<td>Honeywell LSZ-860 Lightning Sensor System</td>
<td>The Certification Basis for this original product remains applicable to this certificate/ approval.</td>
<td>FAA Statement of Compliance letter 1900-13-030, dated 19/02/2013 - Aircraft FlightManual Supplement GC16250M001, Revision- dated1/11/2012 - Instructions for Continued Airworthiness GC16225AM01, Revision 1, dated 27/09/2013 or later revisions of the above listed documents approved by EASA in accordance...</td>
<td>Prior to installation of this design change, it must be determined that the...</td>
<td>22/02/2013</td>
<td>Active</td>
<td>10043755</td>
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<tr>
<td>10043757</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC</td>
<td>ST04259AT-D</td>
<td>GVI (G650)</td>
<td>Honeywell AIS-2000 Satellite TV System</td>
<td>The Certification Basis for this original product remains applicable to this certificate/ approval.</td>
<td>FAA Statement of Compliance letter 1900-13-030, dated 19/02/2013 - Aircraft FlightManual Supplement GC16107M001, Revision- dated 06/10/2012 - Instructions for Continued Airworthiness GC16157A001, Revision- dated 06/10/2012 or later revisions of the above listed documents approved by EASA in accordance...</td>
<td>Prior to installation of this design change, it must be determined that the...</td>
<td>22/02/2013</td>
<td>Active</td>
<td>10043757</td>
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<td>STC Holder</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>S550 (CITATION SIII), 550 (CITATION V), 550 (CITATION ULTRA), 550 (CITATION II), 550 (CITATION I), 550 (CITATION )</td>
<td>10043783</td>
<td>Installation of Universal UHS-1sur FMS System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated criteria noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Scandinvanian Avionics, AFM Supplement 2080144-AM-A, issue 2 dated 23 January 2013; scandinavian Avionics, Continued Airworthiness 2080144-CA-A, issue 1 dated 13 December 2012; scandinavian Avionics, Maintenance Manual Supplement 2080144-AM-A, issue 1 dated 13 December 2012; scandinavian Avionics, Installation...</td>
<td>See Scandinavian Avionics, AFM Supplement 2080144-AM-A prior to installation of this design change it must be determined that the interrelationship between this design change and anything theretofore installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>22/02/2013</td>
<td>Active</td>
<td>10043783</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>TB 20</td>
<td>10043783</td>
<td>Installation of a Garmin GNS 530W system, a GPS GA 35 antenna and a King KN 62A DME</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the product/approval CS-23, Amendment 2. The certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate...</td>
<td>Flight Manual Supplement No AE1208784VGNS 530W, revision 0 dated 25/02/2013; Installation Instructions No AE1208785TB, revision 0 dated 02/08/2012; Instructions for Continued Airworthiness No AE1208785PM, revision 0 dated 25/02/2012; Master Data List No AE1208786MDL, revision 0 dated 25/02/2012 or later revision...</td>
<td>This installation is limited to airplane TB20 SN 1088 and it is approved for IFR B-RNAV operations with coupling to a King KFC 150autopilot and to an Aspen EFIS 1000 Pro display; Software upgrades introducing new functions at equipment level must be subject of other EASA approvals. Prior to installation...</td>
<td>25/02/2013</td>
<td>Active</td>
<td>10043783</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>TB 20</td>
<td>10043786</td>
<td>Installation of an Aspen Evolution 2000 display system</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the product/approval CS-23, Amendment 2. The certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate...</td>
<td>Flight Manual Supplement No AE1208685SB, revision 0 dated 17/09/2012; Installation Instructions No AE1208686IB, revision 0 dated 17/09/2012; Instructions for Continued Airworthiness No AE1208665CA, revision 0 dated 17/09/2012; Master Data List No AE1208673MDL, revision 0 dated 17/09/2012...</td>
<td>The airplane must be equipped with standby altitude, airspeed, attitude, heading and navigation indicators independent from and not similar to the Aspen primary display; Software upgrades introducing new functions at equipment level must be subject of other EASA approvals. Prior to installation of this...</td>
<td>25/02/2013</td>
<td>Active</td>
<td>10043786</td>
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<tr>
<td>AERODATA AG</td>
<td>HAWKER 500</td>
<td>10043842</td>
<td>Installation of AD-APS-355 consisting of operator console and equipment installation for Flight Inspection System (FIS) with associated antennas and installation of camera tube (GHT) in the tail cone. Along with associated modifications to cockpit and electrical interfaces.</td>
<td>The Certification Basis applicable to this certificate/approval is defined by CS-25 Amendment 11 excluding Subpart H and with the following revisions: FAR-25 Amendment 0 defined on TCDS IM.A.085-25.785 FAR-25 Amendment 20 as defined on TCDS IM.A.085-25.789 FAR-25 Amendment 20 voluntarily; The requirements for...</td>
<td>Installation of Aerodata AD-FIS-355 under modification H125-3.01 as defined in Aerodata MDL/AC-185 Rev 2 dated 15 February 2013; Operation in accordance with Aerodata Flight Manual Supplement SFM-13 Rev 0 dated February 2013; Maintenance in accordance with Aerodata Aircraft Maintenance Manual...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third party...</td>
<td>28/02/2013</td>
<td>Active</td>
<td>10043842</td>
<td></td>
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<tr>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td>114, 114A, 114B, 114TC, 112, 112B, 112TC, 112TCA</td>
<td>Installation of HD Xenon LDG/TAXI and LED NAV Lights Replacement of existing installation with new HD and LED installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated criteria noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List, Nayak GmbH, document RC114-11102, rev. 1, dated January 2, 2013; Instructions for Continued Airworthiness and Installation according to Nayak GmbH, document RC114-11102, rev. 0, dated October 26, 2011; Aircraft Flight Manual Supplement, Nayak GmbH, document RC114-11102, rev. 0...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10043862</td>
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**List of EASA Supplemental Type Certificates**

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<th>STC No.</th>
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<th>Foreign/NAA STC ref.</th>
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<td>10043855</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
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<td></td>
<td>MU-2B-35, -36, -40, -40A, -60, MU-2B-20, -25, -28, 28A, -30.</td>
<td>Installation of 5-blade MT-Propeller Model MT-27.1-E-C-F(R)-(G)CFPR.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The following paragraph(s) at a later amendment/FAR 23.1529 Amdt. 28C5c-9 Amdt. 3 and FAR 36 Amdt. 28This certificate/approval involves change to the requirements for environmental protection or as...</td>
<td>According to MT-Propeller Master Document List No. E-2247 and E-2306, or later Revisions; To be operated in accordance with MT-Propeller AFM-S Document No. E-2247 and E-2306; To be maintained in accordance with STC-009(1) dated 04 Feb 2006 (inclusive) and MT-Propeller ICA Document No. E-2247 and E-2307.</td>
<td>28/02/2013</td>
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<td>10043884</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td></td>
<td>B747-8</td>
<td>NG Stretcher installation in aft area of the main deck</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements D-103 &quot;Installation of medical stretcher in cabin&quot; The requirements for environmental protection and the associated certified noise and/or emissions levels of the original...</td>
<td>Conditions 1. All equipment (stretcher, oxygen bottles, etc.) must be approved/against the appropriate qualification standard, if any. 2. The stretcher shall provide an adequate restraining means for the person, using such devices as a shoulder harness, the appropriate number of body-belts and/or end bars...</td>
<td>01/03/2023</td>
<td>Active</td>
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<tr>
<td>10043876</td>
<td>BERINGER AERO</td>
<td>REV. 10</td>
<td></td>
<td>DA-40 NG, DA-40 F, DA-40 NG, DA-40 D, DA-40 F, DA-40 D</td>
<td>Original Issue: STC-009(1) dated 04 Feb 2013 – Installation of RF-006 (nose wheel with EA-004 brake system on DA40NG Revision 6; Update applicability to cover all versions of DIAMOND DA40 Revision 7; Oak – 1/4&quot; fitting screws + MP-001,5,DA40NG – 8i fitting screws replaced by 1/4&quot; screws &amp; intro...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to...</td>
<td>Modification file: DM-SC-09/rev 10 dated 11 Mar. 2022; Installation and Instructions for Continued Airworthiness; ICA - STC-009(1) dated 19 Oct. 2021 or later revisions of the above listed documents approved by EASA.</td>
<td>18/03/2022</td>
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<td>10043917</td>
<td>RHEINLAND AIR SERVICE GMBH</td>
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<td></td>
<td>MT-7-235</td>
<td>Installation of EDM(30)Installation of J.P. Instruments (JPI) Primary Engine Data Management System in accordance with JPI Installation Manual for the EDM 909/930, Report 908, dated September 9, 2004.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - the following paragraph(s) at a later amendment/CS 23 Amendment 3 for...</td>
<td>This approval shall be limited to aircraft MT-7-235 MSN 18087C. Prior to installation of this design change it must be determined that the...</td>
<td>05/03/2013</td>
<td>Active</td>
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<td>10043924</td>
<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U.</td>
<td>REV. 1</td>
<td></td>
<td>MBB BK117 B-2</td>
<td>Adaptation of FLIR operator desk and seat in order to operate the FLIR with the long range fuel tanks (TC holder or optional installation mod), the after seats have to be moved forward and modified to fit in the new attachment points. Also the operator console is redesigned to fit in the available...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement CMA411 - FMS RIMaster document list CMA411 - LMD RIMaster Instruktions for Continuous Airworthiness CMA411 - ICA RIMaster revisions of the above listed document(s) approved/accepted by EASA.</td>
<td>Only applicable to s/n 7176 Prior to installation of this change/repair it must be determined that the...</td>
<td>01/02/2019</td>
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<td>TAMARACK AEROSPACE GROUP, INC.</td>
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<td>FAA STC: SA02475E</td>
<td>SR20, SR22, SR22T</td>
<td>Cargo Conversion Removal of rear seats and installation of cargo shelf and flight/night system in accordance with FAA STC SA02475E.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Removal of rear seats and installation of cargo shelf and flight/night system in accordance with FAA STC SA02475E.</td>
<td>This STC is applicable to the following serial blocks of Cirrus Designaircraft: - Cirrus SR20: Serial Numbers 1005 and subsequent. - Cirrus SR22: Serial Numbers 0002 and subsequent. - Cirrus SR22T: Serial Numbers 0001 and subsequent. Prior to installation of this design change it must be determined that...</td>
<td>05/02/2013</td>
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<td>10043927</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td>AS 365 N3</td>
<td>White Strobe Light Installation on the Underside Fuselage</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The following paragraph (s) at a later amendment: CS 29 AmI.2 paragraphs 29.1, 29.15, 29.29, 29.601, 29.605, 29.607, 29.609, 29.615, 29.777, 29.1301, 29.1309, 29.1351, 29.1353, 29.1357, 29.1359, 29.1361, 29.1363, 29.1365, 29.1367, 29.1369 &lt;;... D3A34C444/78MLA-0</td>
<td>- Modification limited to AS365N3 with Serial Numbers SN6687 and SN6690. - The use of the white strobe light system with Night Vision Imaging Systems (NVIS) or Night Vision Glasses (NVG) is forbidden. - The operation of the white strobe light on the underside in cloudy conditions is forbidden. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the original product as amended by this STC.</td>
<td></td>
<td>06/03/2013</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td>AS 365 N3</td>
<td>Passenger Address System Installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The following paragraph (s) at a later amendment: CS 29 AmI.2 paragraphs 29.1, 29.15, 29.29, 29.601, 29.605, 29.607, 29.609, 29.615, 29.777, 29.1301, 29.1309, 29.1351, 29.1353, 29.1357, 29.1359, 29.1361, 29.1363, 29.1365, 29.1367, 29.1369 &lt;;...</td>
<td>Master Certification Data Sheet, document no. 2026-01-A-2115-R03, Revision 03, Instructions for Continued Airworthiness, document no. 2026-01-A-2211-R02 Revision 02 Service Bulletin, document no. 2026-01-A-3017 R02, Revision 02 or later approved revisions of the above listed documents.</td>
<td>Modification limited to AS365N3 with Serial Numbers SN6687 and SN6690. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the original product.</td>
<td>06/03/2013</td>
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<td>10043931</td>
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<td>MCCAULEY PROPELLER SYSTEMS</td>
<td></td>
<td>LC42-550F</td>
<td>Installation of McCauley Model D3A34C44478MLA-0</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection and a change to the certified noise or emissions levels.</td>
<td>According to McCauley Installation Instructions Doc. No. 999 Rev -., or later Revisions To be operated in accordance with McCauley APM-5 Document No. 970, RevR1 to be maintained i.a.w. McCauley ICA Document No. 975, Rev A or later revisions of the above listed documents approved by EASA in accordance with...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>06/03/2013</td>
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<td>10043935</td>
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<td>ARMSTRONG AEROSPACE Inc.</td>
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<td>AA30-341, A30-342, A30-343, A30-321, A30-322, A30-323, A30-301, A30-302, A30-303, A30-305, A30-233, A30-243, A30-201, A30-202, A30-203</td>
<td>Certification of Compatibility with Cellular and Wireless Data Technologies</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Armstrong Aerospace Master Data List DC-104-6029-00, Revision B dated 14 December 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/47CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EUSA B...</td>
<td>1) This STC does not constitute installation approval of aeroplanehardware. It only provides data that may partially support the installations or activations. This STC may only be applied to an aircraft when an airplane installed base station using the cellular and/or wireless technologies is consonant...</td>
<td>06/03/2013</td>
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<td>EASA STC No.</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10043964</td>
<td>MT-PROPELLER ENTWICKLING GmbH</td>
<td></td>
<td></td>
<td>DHC-6 SERIES 400 DHC-6 SERIES 300, 310, 320 DHC-6 SERIES 200, 210 DHC-6 SERIES 1 100, 110</td>
<td>Installation of 4-blade MT-Propeller Model MTV-16-E-C-F-R (PVCFR240.55m) Viking Air Limited DHC-6</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>According to MT-Propeller Master Document List No. E-2291 and MT-Propeller Installation Instructions Doc. No. E-2294, this modification is to be operated in accordance with MT-Propeller AIPs Document No. E-2200. Its impact upon the airworthiness, safety or emissions of the original product remains unchanged and remains applicable to this certificate/approval.</td>
<td>09/02/2013</td>
<td>Active</td>
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<td>10043965</td>
<td>MT-PROPELLER ENTWICKLING GmbH</td>
<td>REV. 1</td>
<td></td>
<td>GA8-TC 320 GA8</td>
<td>MTV-6-B/200-58 on GA8 Airvan Model GA8 and GA8-TC 320</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>According to MT-Propeller Master Document List No. E-2217 (Rev.3) GA8 Airvan Model GA8 to be operated in accordance with MT-Propeller AIPs Document No. E-2642 GA8 Airvan Model GA8-TC 320 to be operated in accordance with MT-Propeller AIPs Document No. E-2241 for later revisions of the above listed documents/approval.</td>
<td>09/04/2015</td>
<td>Active</td>
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<td>10043973</td>
<td>STANDARD AERO (ALLIANCE), Inc.</td>
<td>FAA STC ST01786CH-D</td>
<td>A330-322/341/342-343 A330-301/302-303/321-322</td>
<td>Activation for the ARINC Cabin Connect System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- FAA Approval ST01786CH-D, dated 14/02/2013.</td>
<td>Before the installation of this STC shall be applied the following STCs: 10043526 FAA Approval Document No. E-2242 for Installation Instructions Document List No. E-2217 FAA STC ST02177-L, dated 08/03/2013.</td>
<td>08/03/2013</td>
<td>Active</td>
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<td>10043975</td>
<td>REIMS AVIATION INDUSTRIES</td>
<td>TCCA SA12-74</td>
<td>BD-700-1A10, BD-700-1A11</td>
<td>Installation, Provisions for P/N 444-DMC57-1 Securaplane Antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Transport Canada STC No. SA12-74, dated 24 October 2012; ACS-NAI Ltd. Modification Data Summary number H791000 Revision N/C Mandatory Maintenance Manual Supplement H791070; revision N/C or later revisions of the above listed documents/approval.</td>
<td>Prior to installation of this modification it must be determined the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/03/2013</td>
<td>Active</td>
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<td>10043977</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST02177LA</td>
<td>CL-600-2B16 (804 VARIANT)</td>
<td>HUD/EVS Model HGS-9605 Validated on CL-605 and CL-650 aircraft. This revision is to include marketing/common designation of the CL-650 on this STC in line with the Bombardier CL-600 TCDS IM-A.013.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- FAA STC ST02177-L, dated 07 Dec. 2015.</td>
<td>- FAA STC ST02177-L, dated 07 Dec. 2015. Rockwell Collins Master Drawing List for Model 6605 Head-Up Display System, Doc. No. 9701-1267, Rev. P, dated 18th July 2016; - Rockwell Collins Master Flight Manual Supplement (AFMS) for Installation of HUD on HUD w/EFVS.</td>
<td>The performance of the Model 6605 HUD system has demonstrated acceptability as supplementary display of primary information. The EVS image does not degrade the safety of flight or interfere with the required flight tasks during any phase of flight. Only the configuration is...</td>
<td>16/03/2018</td>
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<td>Limitations/Conditions</td>
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<td>FAA STC SR02135NY</td>
<td>A119/AV/119MKI</td>
<td>Pilot and Crew Doors removed. The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- Robonaut Flight Manual Supplement Reports No. AAC-ENG-04-048 Rev. B (FAA approved on June 30, 2005) for the A119 model AAC-ENG-08-084 Rev. - (FAA approved on March 4, 2005) for the A119 MKII models. Later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 20...</td>
<td>None</td>
<td>08/04/2013</td>
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<td>FEDERATION FRANCAISE DE VOL A VOILE (FFVV)</td>
<td>10044028</td>
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<td>The Certification Basis for the original product remains applicable to this certificate approval. The Certification Basis for the original product remains applicable to this certificate approval.</td>
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<td>The Certification Basis for the original product remains applicable to this certificate approval. The Certification Basis for the original product remains applicable to this certificate approval.</td>
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<td>13/03/2013</td>
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<td>DASSAULT FALCON SERVICE</td>
<td>10044032</td>
<td>REV.</td>
<td></td>
<td>MF50</td>
<td>Previous 2 main FMS UNS-1Lx with integrated GPS are replaced by two new FMS UNS-1Lx (SCN 1000.X) with X.2 16s with integrated GPS / WAAS. The receiver A third one with deactivated integrated GPS receiver can be installed (Version 2).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Flight Manual Supplement EFD 1000, revision B dated 04/03/2013. Flight Manual Supplement n° AE12079SMV FDN 460, revision 0 dated 04/03/2013. Instruction to continued aeromedical n° AE12079WA revision 1 dated 03/10/2013. Master Data List n° AE12079MDL revision 2 dated 04/03/2013</td>
<td>19/01/2015</td>
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<td>DASSAULT FALCON SERVICE</td>
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<td>MF50</td>
<td>P-RNAV Capability. To demonstrate compliance with TGL N° 10 Rev 1 concerning P-RNAV/operations &quot;Aeromedico and operational approval for Precision RNAV operations in designated European airspace&quot;</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>DFS Change Approval Sheet DF55667-CAS Issue C dated January 13th, 2015. DFS Airplane Flight Manual Supplement N° &quot;F50.40 dated March 8th, 2013.or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce noise/avexeffect upon the aeromedico of the product.</td>
<td>13/03/2013</td>
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<td>10044034</td>
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<td>MF50</td>
<td>COMPLIANCE WITH AC 90-10A To demonstrate compliance with the RNAV 1 and RNAV 2 regulations described in AC90-10A with the document 8613 AN55/3 &quot;Performance-based Navigation (PBN) Manual&quot; for Falcon 50 aircraft equipped with FMS UNS-1Lx software 1000.x (X.2 16s) according to DFS 5687 Issue A or after.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>DFS Change Approval Sheet DF55668-CAS Issue A dated March 8th, 2013. DFS Modification Sheet DF55668, Issue A dated March 8th, 2013. DFS Airplane Flight Manual Supplement N° &quot;F50.43 dated March 8th, 2013.or later revisions of the above listed documents approved by EASA</td>
<td>As per DFS Change Approval Sheet DF55665-CAS Issue B dated March 8th, 2013. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce noise/avexeffect upon the aeromedico of the product.</td>
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<td>DASSAULT FALCON SERVICE</td>
<td>MF50</td>
<td>TA12-20</td>
<td>RNP-10 and RNP-4 capability to demonstrate compliance with the RNP-10 and RNP-4 regulations for Falcon 50 aircraft equipped with FMS UNS-1Lw software 100.X (dot)accor to DFS 5687 issue A or after.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>DFS Change Approval Sheet DFS5689-CAS Issue A dated March 08th, 2013. DFS Modification Sheet DFS5689, Issue A dated March 08th, 2013. DFS Airplane Flight Manual Supplement N &quot;FS-64 dated March 08th, 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>As per DFS Change Approval Sheet DFS5689-CAS Issue A dated March 08th, 2013. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/energy effect upon the a...</td>
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<td>DASSAULT FALCON SERVICE</td>
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<td>TA12-20</td>
<td>RNP APP Capability/Description as per Associated Technical Documentation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>DFS Change Approval Sheet DFS5689-CAS Issue A dated March 08th, 2013. DFS Modification Sheet DFS5689, Issue A dated March 08th, 2013. DFS Airplane Flight Manual Supplement N &quot;FS-65 dated March 08th, 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>As per DFS Change Approval Sheet DFS5689-CAS Issue A dated March 08th, 2013 and DFS Airplane Flight Manual Supplement N &quot;FS-65 dated March 08th, 2013. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed...</td>
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<td>10044038</td>
<td>ROW 44, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST2904OH</td>
<td>Installation of a Row 44 Wireless Broadband System Provisions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>DFS Change Approval Sheet DFS5689-CAS Issue A dated March 08th, 2013. DFS Modification Sheet DFS5689, Issue A dated March 08th, 2013. DFS Airplane Flight Manual Supplement N &quot;FS-66 dated March 08th, 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/energy effect upon the airworthiness of the product.</td>
<td>15/09/2015</td>
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<td>10044049</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 4</td>
<td>FAA STC ST022515E</td>
<td>Installation of a Row 44 Wireless Broadband System Provisions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>DFS Change Approval Sheet DFS5689-CAS Issue A dated March 08th, 2013. DFS Modification Sheet DFS5689, Issue A dated March 08th, 2013. DFS Airplane Flight Manual Supplement N &quot;FS-67 dated March 08th, 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/energy effect upon the airworthiness of the product.</td>
<td>12/12/2013</td>
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<td>10044051</td>
<td>B/E AEROSPACE LIMITED</td>
<td>REV. 1</td>
<td>TCCA STC SA12-40</td>
<td>Installation, Complete Custom Aircraft Interior/Revision 1 is an administrative reissue following a name change fromACS-NAI Limited to B/E Aerospace Limited.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements/ Special Condition(s): (a) Installation of Inflatable Restraints25.307(a) - @ Amtd 125.36(b)(g) - @ Amtd 1325.75(a)(b) - @ Amtd 125.61(b)(ii) - @ Amtd 125.75(b) - @ Amtd 125.61(b)</td>
<td>Master Data List, Doc. No. HPO-CL-158, Rev. V, Dated 03.08.2013CA, Doc. No. HTM1010, rev. F, Dated 10/07.2015 - revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/energy effect upon the airworthiness of the product. The installation of this modification by third p...</td>
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<td>MONARCH AEROSPACE LLC</td>
<td>MONARCH STC-215 Rev 1</td>
<td>1</td>
<td>CAAR-215</td>
<td>A321-201-214/232/233</td>
<td>Installation of the Wireless IFE System</td>
<td>The Certification Basis for the original product as amended by the following or alternative airworthiness requirements: CS 25.1709</td>
<td>Monarch Aerospace Services Flight Manual Supplement MO3A-39-306 No. 1116.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/02/2015</td>
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<td>AIRCRAFT CABIN MAINTENANCE GmbH</td>
<td>EASA STC no. 10044062</td>
<td>1</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of refurbished Passenger Seat Belt Skin EPA 27511.2-01-70 on 16 g seats</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Refer to Master Document List E00 No. 112-100311 Rev. 0 (27.02.2013) or later revisions of the above listed documents approved by EASA</td>
<td>As per Service Bulletin No. 112-100311 SEB 06.0, latest revision. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/03/2013</td>
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<td>AIRCRAFT CABIN MAINTENANCE GmbH</td>
<td>EASA STC no. 10044063</td>
<td>1</td>
<td>FAA STC SA03685CH-D</td>
<td>Installation of an A/C Systems Air-conditioning System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Standard Aero Master Drawing List 1012176 Rev. 0, dated 20 April 2012 Operation in accordance with FAA Approved Airplane Flight Manual Supplement number FTA-010-2 Rev. (I) dated 25 April 2012. Maintain in accordance with A/C Systems LLC ICA and MM Supplement ref FTA-010-2 Rev. (I).</td>
<td>The installer must determine whether this design change is compatibly previously approved modifications. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will intro...</td>
<td>14/03/2013</td>
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<td>AIRCRAFT CABIN MAINTENANCE GmbH</td>
<td>EASA STC no. 10044069</td>
<td>1</td>
<td>FAA STC SA02578B-02</td>
<td>Installation of an A/C Systems Air-conditioning System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with A/C Systems Master Drawing List FTA-001-9 Rev. C, dated 07 June 2012. Operation in accordance with FAA Approved Airplane Flight Manual Supplement number FTA-010-2 Rev. (I) dated 25 April 2012. Maintain in accordance with A/C Systems LLC ICA and MM Supplement ref FTA-010-3 Rev. C</td>
<td>Model F33A Applicability Series CE-748, CE-772 and after Model A36 Applicability Series E-1111, E-1241 through E-5835 and all Modifications. Model G38 All Series. The installer must determine whether this design change is compatibly previously approved modifications. Prior to installation of this design...</td>
<td>15/03/2013</td>
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<td>KUERZI AVIONICS AG</td>
<td>EASA STC no. 10044092</td>
<td>1</td>
<td>FAA STC SA02578B-02</td>
<td>Installation of an A/C Systems Air-conditioning System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the previous product unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement Doc. No. MFS-200711748 Rev. 0-Nor later revisions of the above listed documents approved by EASA Master Document List Doc. No. MDS-200711748 Rev. C, Instructions for Continued Airworthiness Doc. No. ICA-200711748 Rev. C</td>
<td>Refer to applicable RFM for list of equipment mandatory for flight. Visomage System certification. Installation of NVIS/ NVG compatible lighting does not include or imply approval for flight operations with Flight Vision Goggles (NVG). The operator must receive approval from the Civil Aviation authority...</td>
<td>18/03/2013</td>
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<td>ALAMO ENGINEERING GmbH</td>
<td>EASA STC no. 10044141</td>
<td>1</td>
<td>FAA STC SA02578B-02</td>
<td>Installation of an A/C Systems Air-conditioning System</td>
<td>The Certification Basis for the original product as amended by the following or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.1709 is CS 25.1729 at Amendment 11. The requirements for environmental protection and the associated certified noise...</td>
<td>Declaration of Compliance Doc. No. DC-10030232001, Issue Baladeb.03.2013 Master Document List, Doc. No. MDS-200712001, Issue A dated 19.03.2013. Instructions for Continued Airworthiness Doc. No. ICA-200712001, Issue A dated 19.03.2013. AFM Supplement Doc. No. AFMS01004320001, Initial Release date...</td>
<td>Aircraft Serial Number 7617 only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/12/2013</td>
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<td>AVIONIK STRAUSSING ENTWICKLUNGS GmbH</td>
<td>Installation of Garmin GTS820 with GPA955 and Dual Garmin GMA84, GarminGDL89, DAC GDC31, Garmin GA36, Garmin GAA7 and Comart CI 490-1</td>
<td>- Flight Manual Supplement Garmin GTS820 ASR-2013-015, FMS-05-1-00, Rev. 0. Flight Manual Supplement GDC31 ASR-2013-015, FMS-05-2-00, Rev. 0</td>
<td>None</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>STC Twenty One Limited</td>
<td>Upgrade To Existing VIP Interior. This Rev: 1 is to amend a typographical error in &quot;Des...</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>Installation of Airplane Completion/Revision to address novel and unusual design features associated with shower installation</td>
<td>- FAA STC ST-04261AT-D, issued 27th June 2012. - Gulfstream Aerospace Corporation Index List GC5210050365, Revision F, dated 14th November 2013 - GV-SP Airplane Flight Manual Supplement GC512040M006, Revision D, dated 14th November 2013. - Instruction for Continued Airworthiness GC528011785, Revision D, 4...</td>
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**List of EASA Supplemental Type Certificates**

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An agency of the European Union
### List of EASA Supplemental Type Certificates

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<th>Date of Issue</th>
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<td>TENENCIA Ltd.</td>
<td>10044218</td>
<td>REV. 1</td>
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<td>AB 412 EP</td>
<td>EMS Configuration as per Master Drawing List MDL-25-00-1213-A, revision 01, dated 22 March 2013. Rev. 1 is an administrative re-issuance to reflect the recent namechange from AINER Aviation Italia S.p.A. Divisione Ala Fissa Rev. 2 is...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness ofthe product.</td>
<td>26/03/2013</td>
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<td>AVIONTEC Ltd.</td>
<td>10044281</td>
<td>REV. 1</td>
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<td>P210N</td>
<td>Removal of: Siptronics SPA400 Intercom; King KMA 24 Audio panel; King IX 165 NAV/COM; King KLN90E; GPD Installed; Garmin GMA 350 Intercom/Audio panel; Garmin GNS 430W NAV/COM/GPS/L-1900S Stormscope/Compliance showing a MIL-STD-2528 that the installed equipment is suitable for RNAV (GNSS) Approached...</td>
<td>Master Document List Doc. No. 2259-017, dated 20.02.2013 Instructions for continued airworthiness: Doc. No. 2259-015, dated 20.02.2013 Instructions for continued airworthiness: Doc. No. 2259-011, dated 20.02.2013</td>
<td>N/A</td>
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<td>10/04/2013</td>
<td>Flight Manual Supplement BK17-02441-025-MFS Rev. 0 or later revisions of the above listed documents approved by EASA-Master Document List BK17-02441-001-LMD Rev. 0. Certification Programme BK17-02441-003-RCP Rev. 1. Installation Instructions BK17-02441-004-DIN Rev. 0. - Instructions for Cont..</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
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<td>10/04/2013</td>
<td>Master Document List, Doc. No. MDL-12-005 Rev. E dated May 04, 2003; Certification Program, Doc. No. CPR 10-057 Rev. B dated Apr 25, 2003; Flight Manual Supplement, Doc. No. FMS-429-001190-001 Rev. B dated Sep 01, 2021 Instructions for Continued Airworthiness included in Operations Manual Doc. No. OPM.</td>
<td>The Type and OISD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remain applicable to this certificate/approval.</td>
<td>S/N 57067 and S/N 57330 only. The medical portable devices, including their operation, have to be approved and compatible with the helicopter. If the operator for electromagnetic interference (EMI) and radio frequency interference (RFI) The installation of this modification by third p..</td>
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<td>05/04/2013</td>
<td>Master Document List Doc. No. MDL-200931669 Rev. NIC of 03/04/2013 for Cont Airworthiness ICA-200931669 Rev. NIC of 03/04/2013 for later revisions of the above listed documents approved by EASA.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certification noise/ or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>S/N list Prior to installation of this modification it must be determined that the installation of this modification and any other previously installed modification/repair will introduce no adverse effect on the airworthiness of the product.</td>
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<td>10044362</td>
<td>08/04/2013</td>
<td>Flight Manual Supplement FMS-350-24-101-01 Rev. for later revisions of the above listed documents approved by EASA-AirRv A S Master Document List DO-350-24-101 Rev. 0 dated 14 February 2013 Instruction s for Continued Airworthiness ICA-350-24-1-01-01 Rev 1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change/repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>Active</td>
<td>10044362</td>
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<td>10044402</td>
<td>10/04/2013</td>
<td>MDL HC20100500 Rev N dated 25/06/2004-4FM supplements FM2585002 Rev NC dated 15/12/2003 and H- FM2585003 Rev NC dated 25/06/2004-4ICAO20153000-IFCA-1or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/044/CF (or subsequent revisions offset this dec..</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (Equivalency Safety Findings) O-08 issue 3 dated 03/04/2013 Deviation(s) D-07 issue 3 dated 3/04/2013D-09 issue 3 dated 3/04/2013D-10 issue 3 dated 3/04/2013The requir..</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change/repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>Active</td>
<td>10044402</td>
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<td>10044405</td>
<td>27/04/2021</td>
<td>EASA Master Data List No. AES-767-2349 MDL issue 3 dated 04 April 2013AES Instruction for Continued Airworthiness No. AES-TR-1062 Issue 4 dated 24 June 2021 Instructions for Continued Airworthiness ICA-767-2349-1 Issue 3 dated 24 June 2021</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (Special Condition (s) CRI D-01 Installation of Crew Rest Compartment(s) C-25 Amendment 11 - Various The requirements for environmental protection and the associated certified fueld..</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change/repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>Active</td>
<td>10044405</td>
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<td>STC Holders</td>
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<tr>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>Installation of 40 Cubic Foot Oxygen Bottle.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unachieved and remain applicable to this certificate/ approval.</td>
<td>FAA STC No. ST0254CH-D dated 30 January 2013. Jet Aviation St. Louis, Inc. Master Data List 41-261G001, Revision G dated 30 November 2012. Airplane Flight Manual Supplement 41-261M001, Revision NC, dated 4 June 2008 Instructions for Continued Airworthiness 41-027E019 or later revisions of the above...</td>
<td>Restricted to ASN 5701 and subsequent. If this STC is installed in conjunction with the Cabin Interior STC No EASA.4.A.9.02530 (FAA STC No. ST0254CH-D), the Oxygen system Airplane Flight Manual Supplement from the STC EASA.4.A.9.02530 must be replaced with Airplane Flight Manual Supplement 41-261M001.</td>
<td>15/04/2013</td>
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<td>1004441</td>
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<tr>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES.A.R.L</td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative anechoic test requirements applicable to this STC:</td>
<td>Supplement au manuel de vol EDM700 Rev. 1 du 28 janvier 2013.Supplement au manuel de vol GNS 430W Rev. 1 du 28 janvier 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>The STC is approved for aircraft equipped with automatic pilot equipment defined in GARMIN IM 100-00356-08 dated May 2011 and up and STC 4022/12 annex C2.Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change is maintained.</td>
<td>11/04/2013</td>
<td>Active</td>
<td>1004442</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>Installation of S-Tec system 60 Two Axis Autopilot model ST-278,according to bulletin N 376 rev 1 dated 3-Oct-1986 (FAASTC S80372SW-D).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unachieved and remain applicable to this certificate/ approval.</td>
<td>According to S-Tec Master Drawing List N. 93231 rev. A dated 3-Oct-1986 with optional Flight Director/ Steering Horizon;To be operated in accordance with S-Tec AFM/S-P N 89/86 dated 11- Feb-1985,To be operated in accordance with S-Tec AFM/S-P N 89/86 dated 10-05-1986,To be maintained i.a.w. Instruction...</td>
<td>The installation of this modification by third persons is subject to the approval of the STC holder.</td>
<td>12/04/2013</td>
<td>Active</td>
<td>1004442</td>
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<tr>
<td>S-Tec Twenty One Limited</td>
<td>Installation of VHF Comm &amp; Satcom Antennas.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unachieved and remain applicable to this certificate/ approval.</td>
<td>STC Twenty One Limited Master Document List No. S21.MDL-0099 InitialIssue dated 27 2013.Instructions for Continued Airworthiness Supplement Manual No.S21.TEC-0370 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change is maintained.</td>
<td>15/04/2013</td>
<td>Active</td>
<td>1004445</td>
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<td>MAGNETIC MRO AS</td>
<td>Installation of Aivo economy and business class seats with Lumexis FTTSIFE system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unachieved and remain applicable to this certificate/ approval.</td>
<td>- Certification Programme: 110P117, issue 4 (18 March, 2013)-Compliance Checklist: 120C179, issue 2 (March 2013)-Compliance Checklist: 120C179, issue 2 (March 2013)- Compliance Checklist: 120C182, issue A (January 2013) Instructions for Continued Airworthiness: 11CA059, issue C (28 March, 2013).Instruct...</td>
<td>STC applicable for B747-400 aircraft MSN 26637 and MSN 26638, to beoperated with S-Tec AFM/S-P N 89/86 dated 11- Feb-1985.</td>
<td>19/06/2019</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>STC Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>Date of Issue</td>
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<tr>
<td>LAKE CENTRAL AIR SERVICES</td>
<td>REV. 1</td>
<td>TCCA SA08-85</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Instruction Installations are: Eligibility List LCA-SFW-2; Rev. 2 dated 19 July 2010; Instructions for Continued Airworthiness; Doc. No. LCA0010-CD Rev. 0; in later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canada Bilateral...</td>
<td>25/04/2013</td>
<td>Active</td>
<td>10044460</td>
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<tr>
<td>ICELANDAIR ENGIRENING DEPARTMENT</td>
<td></td>
<td>757-200</td>
<td>Communications - Data Transmission &amp; Automatic Calling- DLK of ACARS / CPDLC installation/Installation and activation via Spektral DAH system which provide Aircraft Communications Addressing and Reporting System (ACARS) / Controller Pilot Data Link Communications (CPDLC) with Air Traffic Control...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:...</td>
<td>- Change/Repair Approval, Icelandic Aeronautical Department No. ORA 757-2230-52 Revision C dated 21 March 2013; Continued Airworthiness Instructions, Icelandic document No. CA-757-2230-52; Rev. B, 25 March 2013; Aircraft Flight Manual Supplement, Icelandic document No. AFM-757-2230-52; (more)...</td>
<td>17/04/2013</td>
<td>Active</td>
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<tr>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td></td>
<td>FAA STC ST0121CH</td>
<td>8747-400, 8747-400F</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:...</td>
<td>- ECS-207719, rev. F, Electronic Cable Specialist Master DataLink&gt;(rev. A) &gt;ECS-207678, rev A, Project Compliance Report&gt;(rev. A) &gt;ECS-207719, rev C, AFM Supplement&gt;(rev.) &gt;ECS-207719, rev A, Instructions for Continued Airworthiness&gt;(rev.) &gt;ECS-207719, rev A, ECS-210738 International Airworthiness Certifications Comparison for FAA STC ST0121CH...</td>
<td>27/05/2013</td>
<td>Active</td>
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<td>S-TEC CORPORATION</td>
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<td>FAA STC SA05981C-D</td>
<td>36, A36, F33A</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>- According to S-TEC Master Drawing List N. 92/761 rev.2 dated 15 Jan-2002; To be operated in accordance with S-TEC AFM-S-PN 891156 dated 15 Apr-1995 and AFM-S-PN 891156 rev.1 dated 30 Mar-2001 and AFM-S-PN 891156 rev.2 dated 18 Jan-2002 in respect of the limitations of the certificate,...</td>
<td>17/04/2013</td>
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<td>EASA STC No.</td>
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<td>Rev.</td>
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<td>10044536</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 2</td>
<td>A321-271N/-272N/-273N</td>
<td>A321-271N/-272N/-273N</td>
<td>The Certification Basis for the original product as amended, including any associated technical requirements. The certification basis includes any applicable airworthiness requirements and associated technical requirements. The installation of the Thales ISFD system.</td>
<td>Master Data Unit 0021-012-A2002-T2, Instructions for Continued Airworthiness: 0621-A1-2211, R0985/AMS: 0621-A1-AMF5-R026.</td>
<td>Prior to installation of this design change, the installation must be in accordance with the EASA system.</td>
<td>23/12/2019</td>
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<td>10044546</td>
<td>ICELANDAIR DESIGN ORGANISATION</td>
<td>REV. 2</td>
<td>FAA STC ST03183CH</td>
<td>757-200-300</td>
<td>Installation of the Thales ISFD System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Master Drawing List, Global Aerospace Document No. GR0002-MDL, Revision C. - N10-MD500-369-HV2 ANA: Rotorcraft Flight Manual Supplement - Global Aerospace Document No. GR0003-KFX.</td>
<td>Prior to installation of this design change, the installation must be in accordance with the EASA system.</td>
<td>14/11/2017</td>
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<td>10044546</td>
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<td>10044552</td>
<td>SAFRAN ELECTRONICS &amp; DEFENSEBUSINESS UNIT HELICOPTERS DATA MANAGEMENT SOLUTIONS &amp; SERVICES</td>
<td>REV. 2</td>
<td>369D, 500N</td>
<td>Installation of HELI-COM V2 ANA Embedded UMG Equipment Rev. 1 Extension to model MD369D Rev. 2: STC holder name change from I.S.E.I International Services to SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT HELICOPTERS DATA MANAGEMENT SOLUTIONS &amp; SERVICES.</td>
<td>The Certification Basis for the original product as amended, including any associated technical requirements.</td>
<td>The following additional or alternative airworthiness requirements document(s) approved/accepted under the EASA system.</td>
<td>The installation of this design change must be determined not to introduce adverse effect upon the airworthiness of the original product.</td>
<td>27/02/2020</td>
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<td>10044567</td>
<td>CRANFIELD AEROSPACE LTD</td>
<td>CAA UK RESPONSIBILITY</td>
<td>421C</td>
<td>Structural Provisions of NELIS Antenna Installation in accordance with CAAM Cessna 1360</td>
<td>The Certification Basis for the original product as amended, including any associated technical requirements.</td>
<td>The installation of this design change must be determined not to introduce adverse effect upon the airworthiness of the original product.</td>
<td>19/04/2013</td>
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<td>10044570</td>
<td>STERLING AVIATION TECHNOLOGIES Inc.</td>
<td>PMA APPROVAL PQ169MN</td>
<td>PA-38-112</td>
<td>Approval of PMA PtN SA12001 (replacing P/N 77553-05): Vertical Fin Attach plate between Vertical Fin Forward Spar to a Fuselage Bulkhead (PMA approval PQ2158NM - as per FAA PMA letter dated July 27, 2005).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The approval holder shall fulfill the obligations of Part 21, PMA approval PQ2158NM - as per FAA PMA letter dated July 27, 2005.</td>
<td>The approval holder shall fulfill the obligations of Part 21, PMA approval PQ2158NM - as per FAA PMA letter dated July 27, 2005.</td>
<td>19/04/2013</td>
<td>Active</td>
<td>10044570</td>
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## List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<tr>
<td>10044593</td>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 2</td>
<td>A340-313</td>
<td>TH7 A340 Cabin &amp; IFE Retrofit: The change consists out of Cabin layout: - Removal of all passenger seating and installation of new Business and Economy seats. - Removal of some monuments and installation of new build galley, storages &amp; partitions. - Rewire of Cabin, including ceiling panels and side walls. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: SCD-1322-002 The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List Doc. No. S.0404.000.046- MDL, Rev. 01, dated 15.05.2013. AFMS, Doc. No. S.0404.000.046-AFM01, Rev. 00, dated 25.03.2013. AFMS, Doc. No. S.0404.000.046-AFM02, Rev. 00, dated 28.03.2013. ICA, Doc. No. S.0404.000.046-ICA, Rev. 01, dated 09.04.2013. ICA, Doc. No. C.0404.000.175-ICA, Rev...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/radiation effect upon the airworthiness of the product.</td>
<td>20/04/2021</td>
<td>Active</td>
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<tr>
<td>10044596</td>
<td>AEROSOLUTION GmbH</td>
<td></td>
<td>AW100SP</td>
<td>Quick change patient evacuation system - The installation is identifyed the p/n 109RN5-250001. It consists in an interior modification with EMS. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>109AST-RMS-001-A RPM Supplement later revisions of the above listed documents approved by EASA.</td>
<td>The compatibility of the &quot;Quick Change Patient Evacuation System&quot; therewith previously approved kit must be determined by the installer prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change...</td>
<td>22/04/2013</td>
<td>Active</td>
<td>10044596</td>
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<td>STC Holder</td>
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<td>AIM AVIATION (HENSHALLS) LTD</td>
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<td>FAA STC</td>
<td>10044656</td>
<td>Installation of a Galley and associated ceiling panels into Airbus A330-300 to be operated by Delta.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>AIM Aviation STC Definitive Document No. DD 1209 Issue New dated 05/01/13 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>26/04/2013</td>
<td>Suspendere d</td>
<td>10044656</td>
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<td>CONCORDE BATTERY CORPORATION</td>
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<td>FAA STC</td>
<td>10044684</td>
<td>Replacement of original equipment nickel cadmium batteries with ConcordeModel RG-30E/4L3 Lead Acid batteries.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Concorde Battery Corporation Master Drawing List ref S-0333, Revision B, dated September 15, 2011. Operations in accordance with Concorde Battery Corporation FAA Approved Flight Manual Supplement ref S-0354, Revision B, dated August 15, 2011. Maintain in accordance with Co.</td>
<td>This modification has been reviewed and it has been determined that it does not necessitate a revision to the CWIS ICA that was required to be developed by CR.</td>
<td>29/04/2013</td>
<td>Active</td>
<td>10044684</td>
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<td>AERO MEDICAL PRODUCTS Co. Inc.</td>
<td></td>
<td>FAA STC</td>
<td>10044690</td>
<td>Installation of EMS stretcher system using installation kit 801, including oxygen system 807, inverters and pump kit 812, in accordance with FAA STC SA1130GL.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation and Operating Instructions, Form EA 109K, dated April 23, 2005 Installation and Operating Instructions, Form EA 170, dated June 9, 2001 Aircraft Flight Manual Supplement dated April 18, 2005 Aircraft Maintenance Manual Supplement, Form 107, Rev 2 dated January 15, 2013 or later revisions thereof.</td>
<td>Prior to the first flight of each modified aircraft, and after each change to any of the Communications, Navigational, Identification, or Med Pack equipment, a ground EMI test must be conducted in accordance to EA Form 173, dated November 15, 1993 or later FAA approved revisions. A copy of EA Form 173 must be kept with the Copy of the above listed documents.</td>
<td>29/04/2013</td>
<td>Active</td>
<td>10044690</td>
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<tr>
<td>AIRBUS S.A.S.</td>
<td></td>
<td>FAA STC</td>
<td>10044692</td>
<td>Installation of a Galley and associated ceiling panels into Airbus A330-300 to be operated by Delta.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>As defined in the above listed documents.</td>
<td>This STC is limited to MSN 48225. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a runway effect upon the airworthiness of the product.</td>
<td>12/06/2014</td>
<td>Active</td>
<td>10044692</td>
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<tr>
<td>BE AEROSPACE, Inc.</td>
<td></td>
<td>FAA STC</td>
<td>10044694</td>
<td>Installation of the Cobham HGA-7001 High Gain SATCOM antenna Sub System [DESCRIPTION: Removes and replaces the existing Canadian Marconi CMA-2102 High Gain SATCOM antenna with the Cobham HGA-7001 antenna Revision 3 manufactured and supplied by EMTEQ Engineering to BE Aerospace.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List (MDL); EMTEQ document 2-106975-021, Revision B, dated September 19, 2012 - Instructions for Continued Airworthiness (ICA), EMTEQ document 2-106975-403, Revision C, dated September 19, 2012. &gt;or later revisions of above listed documents approved by EASA.</td>
<td>None</td>
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<td>SPECIALIST AVIATION</td>
<td>MD 902 with PW 206E &amp; 207E</td>
<td>The Certification Basis for the original product as amended by this STC employing additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: CS-27 Amdt.2; 1, 301,303, 305, 307, 311, 337, 341, 561, 601, 603, 605, 609, 610, 611, 613,771, 773, 777, 853, 1301, 1307,1-</td>
<td>Revision 0 - Flight Manual Supplement FMS/669-002 Issue 0 for later revisions of the above listed documents approved by EASA - Design Report DR/669-001 Issue A-Compliance Checklist CCL/669-003 Issue A-ICA MPS/669-008 Issue B-Revision 1. For SAS MOD/800 only - Certification Plan OP/800-001 Issue A- Compliance...</td>
<td>02/03/2018</td>
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<td>BOND HELICOPTERS</td>
<td>Trakea Searchlight on NVIS EC 135</td>
<td>CS 27 Amendment 3The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/ approval.</td>
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<td>None</td>
<td>30/04/2013</td>
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<td>DAO AVIATION A/S</td>
<td>Installation of Electric Flight Display, Aspen EFD 1000 Pro.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>DAO Aviation A/S, Design order Package DAO- DD-0490-DOP-00, dated 20-09-2012 - DAO Aviation A/S, AFM-supplement DAO-DD-0490- AFMS-00, Rev. IR, dated 20-09-2012 - DAO Aviation A/S, Design Report DR/076-001, Issue 1 - Certification Plan OP/076-012, Issue 2</td>
<td>None</td>
<td>30/04/2013</td>
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EASA STC no.: 10044603
STC Holder: SPECIALIST AVIATION SERVICES LTD
Rev.: 1
Foreign/NAA STC ref.: 30/04/2013
Model: MD 902 with PW 206E & 207E
Description: The Certification Basis for the original product as amended by this STC employing additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: CS-27 Amdt.2; 1, 301,303, 305, 307, 311, 337, 341, 561, 601, 603, 605, 609, 610, 611, 613,771, 773, 777, 853, 1301, 1307,1-...
Certification basis: Revision 0 - Flight Manual Supplement FMS/669-002 Issue 0 for later revisions of the above listed documents approved by EASA - Design Report DR/669-001 Issue A-Compliance Checklist CCL/669-003 Issue A-ICA MPS/669-008 Issue B-Revision 1. For SAS MOD/800 only - Certification Plan OP/800-001 Issue A- Compliance...
Limitations/Conditions: Installation of NVIS/NVG compatible lighting does not include or imply approval for flight operations with Night Vision Goggles (NVG).
The Operator must receive approval from the civil aviation operations authority to operate with NVG.Rotorcraft modified by this STC employs Night Vision Imaging System...
Date of Issue: 02/03/2018
Status: Active
System Record: 10044603
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<tr>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>10044719</td>
<td>S550 CITATION S/N 550 CITATION II 550 CITATION BRAVO</td>
<td>Stand-alone AVIATOR 300 Satcom system installation as identified in Engineering Order 202288-EO</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Scandinavian Avionics, AFM Supplement 202288-AM, issue 1 dated 09 July 2012; Scandinavian Avionics, Continued Airworthiness 202280-CA, Issue 2 dated 09 November 2012; Scandinavian Avionics, Maintenance Manual Supplement 202280-AM, Issue 2 dated 14 September 2012; Scandinavian Avionics, Dragon List...</td>
<td>30/04/2013</td>
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<td>NORTWEST AEROSPACE TECHNOLOGIES INC</td>
<td>10044745</td>
<td>A321-231</td>
<td>Cabin interior modifications and In Flight Entertainment installation on Dragon Air.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Plan No.0405-D01 Rev. B, dated 20/02/2013; Certification Plan No.0452-D01 Rev. -, dated 25/10/2013; Certification Summary No.0452-D00 Rev. A, dated 02/05/2013; Certification Summary No.0452-D00 Rev. -, dated 24/03/2014; Master Drawing List N0465-D10 Rev. L, dated 24/03/2014 - Instructions for...</td>
<td>27/03/2014</td>
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<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>10044749</td>
<td>757-200, 757-200PF</td>
<td>Installation of the ACSS TCAS system [description: Installation and activation of ACSS TCAS Change 7 software and deactivation of the ACSS TCAS Change 7 software]</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, Honeywell document No. ML4070737, Revision F, dated 24 July 1990; - Aircraft Flight Manual Supplement, Honeywell document No. 5140-0885, approved on 14 July 1990 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2005/046 and/or...</td>
<td>None</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>10044760</td>
<td>A330-243</td>
<td>Installation of new Avio E-Biz BC Seats, Webey YE Seats, and IMS RaveIFE System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, ref: JN424 ADC-17 Issue 1 Notice for Continued Airworthiness, ref: JN424 ADC-25 Issue 1 Weight and Balance Report, ref: JN424 ADC-20 Issue 1 - for later revisions of the above listed documents approved by EASA, None</td>
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<td>STC Holder</td>
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<td>LEARJET Inc.</td>
<td>FAA STC SAB159NM-D</td>
<td>CL-600-2B16 (CL-604)</td>
<td>Installation of a LH Fwd Storage Cabinet</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. SAB159NM-D dated 29 April 1999. Bombardier Completion Center Master Drawing List No. CA-3079a-TUC; Revision Biladed 26 April 1999 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 200404C (or subsequent revisions off this decision)* and/or.</td>
<td>None</td>
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<td>LEARJET Inc.</td>
<td>FAA STC SAB04HNM-D</td>
<td>CL-600-2B16</td>
<td>Installation of Cabinets and Galley</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC No. SAB04HNM-D dated 03 May 1997. Bombardier Completion Center Master Drawing Lists: .... Revision. Date, C A-30584-TUC, C, ...., 06 February 1999, CA-30555-TUC, C, ...., 06 February 1999, CA-30516-TUC, C, ...., 06 February 1999, CA-30555-TUC, C, ...., 06 February 1999, CA-30555-TUC, C, ...., 06 February 1999, CA-30555-TUC, C, ...., 06 February 1999, CA-30555-TUC, C, ...., 06 February 1999, CA-30555-TUC, C, ...., 06 February 1999, CA-30555-TUC, C, ...., 06 February 1999, CA-30555.</td>
<td>This approval is for installation of structural provisions only. The installation of electrical, plumbing and ECS components must be separately approved.</td>
<td>05/06/2013</td>
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<td>AILIFT A/S</td>
<td>AFW139</td>
<td>Night Vision Image System (NVIS) Certification for AW139 helicopter</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Flight Manual Supplement FMS-139-33-101-01&lt;,&gt;Rev 0 dated 14 March 2013 or later revisions of the above listed documents approved by EASA.- Design Order: DO-139-33-101, Rev 0 dated 3 December 2012- Instruction for Continued Airworthiness: ICA-139-33-101, Rev 0 dated 12 March 2013- Master Doc...</td>
<td>Refer to applicable Flight Manual Supplement (FMS) for list of equipment mandatory for Night Vision Imaging System (NVIS) certification. Installation of NVIS/NVG compatible lighting does not include or imply approval for flight operations with Night Vision Goggles (NVG). The operator must receive approval from EASA.</td>
<td>06/05/2013</td>
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<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC ST0294ICH</td>
<td>B737-600/700/800/900ER</td>
<td>Installation of a Rosemount Aerospace 8700E1 Class 3 Electronic FlightBag (EFB) System. Approved configurations are the following: Configuration 1: Installation of Dual SDU with electrical power and options capable of being configured for continuous operation. Configuration 2: Installation of Dual SDU and EFBB.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Plan, ECS-209047, Revision G, dated 27 April 2011.- Project Compliance Report, ECS-209073, Revision D, dated 02 June 2011. EASA Certification Comparison for STC ST0294ICH ECS-210104, Revision B, dated 04 January 2013- Master Data List ECS-209048, Revision L, dated 13 March 2013. EASA...</td>
<td>Limitations 1) The aircraft operator must perform the actions required for continuous operation as per the Enhanced Flight Bag as listed in the document ECS-209658, Revision A, dated 25 April 2011. 2) Information displayed on the Electronic Flight Bag is not to be used for primary navigation, and shall be used for reference only. 3) The USBA...</td>
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<td>10044818</td>
<td>STC TWENTY ONE LIMITED</td>
<td>ST00057WI</td>
<td>CL-600-2A12, CL-600-2B16</td>
<td>Installation of Left Hand Forward Wardrobe Cabinet</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>FAA STC No. ST00097WI dated 7th October 1996; Learjet Master Document List No. CA-1023-TUC Revision G dated 1993; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Guide.</td>
<td>This approval is for installation of structural provisions only. The installation of electrical, plumbing and ECS components must be separately approved. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed change will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/05/2013</td>
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<td>10044834</td>
<td>LEARJET Inc.</td>
<td>FAA STC No. ST00046WI</td>
<td>CL600-2A12, CL-600-2B16</td>
<td>Installation of Left Hand Forward Wardrobe Cabinet</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>FAA STC No. ST00046WI dated 7th October 2006; Learjet Master Document List No. CA-1023-TUC Revision E dated 2005; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Guide.</td>
<td>This approval is for installation of structural provisions only. The installation of electrical, plumbing and ECS components must be separately approved. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed change will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>LEARJET Inc.</td>
<td>FAA STC No. ST00044WI</td>
<td>CL600-2A12, CL-600-2B16</td>
<td>Installation of Left Hand Forward Wardrobe Cabinet</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>FAA STC No. ST00046WI dated 7th October 1996; Learjet Master Document List No. CA-1023-TUC Revision G dated 1993; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Guide.</td>
<td>This approval is for installation of structural provisions only. The installation of electrical, plumbing and ECS components must be separately approved. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed change will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10044839</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC No. ST01991NY-D</td>
<td>737-800, 737-900, 737-600, 737-700</td>
<td>Installation of Bruce Aerospace cabin lighting</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>Delta Engineering Master Data List 0106-0443-2340 dated March 28, 2013; Instructions for continued airworthiness (ICA) and EWIS: ICA0706-0830-2340 Rev A, applicable to configuration 1 (BR9542 and BR9543) and to configuration 2 (BR9542 and BR9543); or later revisions of the above listed documents appr...</td>
<td>None</td>
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### List of EASA Supplemental Type Certificates

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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tr>
<td>SECUARPLANE TECHNOLOGIES, Inc.</td>
<td>FAA ST0471L-AO</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Installation of Hawker PN 975000270 Series 24VT/4350mp/hour Sealed-Acid Battery</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- FAA ST/STC ST01471L-AO, released 6th October 2004, with FAA Approved Model List (AML) AML-ST01471L-AO at Revision G; FAA Approved Master Data List (MDL) ref. doc. 170.04404.01, Revision Effect Cessna Citation Series Aircraft Certification Data List (CDDL) ref. doc. 170-0411-01, Revision D (for Cessna...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/11/2014</td>
<td>Active</td>
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<td>AEROTECH &amp; CONCEPT</td>
<td>DC-9-83 (MD-83)</td>
<td>Cabin modification and refurbishing. Refurbishing of the cabin and addition of a VIP private salon; updating airframe system; replacement of the path-marking system by a fluorescent one.</td>
<td>The Certification Basis for the original product as amended by the above additional or alternative airworthiness requirements.Special conditions D-01 Glass screen of display-monitor2D-04 Isolated compartmentsD-01 Enhanced airworthiness programme for aero plane systems - ICAS onEWISDeviations:D-02D-03D-05</td>
<td>Master Data List AE12056MDL Rev. 2 dated 08.05.2013 and Rev 2 dated 08.05.2013</td>
<td>AE12056FA Simulation List AE12056MDL, Rev. 0 dated 08.05.2013</td>
<td>Instruction for Continued Airworthiness AE12056ICA Rev. 2 dated 15.04.2013 or later revisions of the above listed documents apply.</td>
<td>DC-9-83 (MD-83) Serial Number 49609</td>
<td>10/05/2013</td>
<td>Active</td>
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<td>H4 AEROSPACE (UK) Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>757-200</td>
<td>Installation of Astrotronics EmPower IPSIS system (Installation of Astrotomics EmPower IPSIS system for European Aviation22C20755 seats into B757-200 aircraft)</td>
<td>The Certification Basis for the original product as amended by the above additional or alternative airworthiness requirements.Special conditions: D-01 Glass screen of display-monitor2D-04 Isolated compartmentsD-01 Enhanced airworthiness programme for aeroplane systems - ICAS onEWISDeviations:D-02D-03D-05</td>
<td>Master Data List AE12056MDL Rev. 2 dated 08.05.2013 and Rev 2 dated 08.05.2013</td>
<td>Instruction for Continued Airworthiness AE12056ICA Rev. 2 dated 15.04.2013 or later revisions of the above listed documents apply.</td>
<td>Modification Summary and Declaration of Compliance, H4 Aeroplanedocument No. HH4-483, Issue 1, dated 10 May 2013- Aircraft Maintenance Manual Supplement, H4 Aeroplane document No.HMMS071, Revision 4, dated 10 May 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>DC-9-83 (MD-83)</td>
<td>13/05/2013</td>
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<td>AVIONICS INTEGRATION &amp; ENGINEERING INC.</td>
<td>LEARJET MODEL 45</td>
<td>5240-049</td>
<td>ISAT Satcom Antenna Installation (Reference: Master data list Doc. No. 5240-049)</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval. JAR-25 Change 13CRM special condition A107The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List 5240-049 dated 06.05.2013 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/05/2013</td>
<td>Active</td>
<td>10044965</td>
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<td>DIAMOND UPGRADIES, L.L.C.</td>
<td>FAA 5N2180AL</td>
<td>412</td>
<td>PART 1: INSTALLATION OF PRATT&lt;... Whitney CANADA PT6T-3DFTWIN-PACPOWER PLANT WITHOUT INCREASED PERFORMANCE CAPABILITY PART 2 INSTALLATION OF PRATT&lt;... Whitney CANADA PT6T-3DFTWIN-PACPOWER PLANT AND INSTRUMENT CHANGE WITH INCREASED PERFORMANCE CAPABILITY. This STC allows for removal of the P&lt;...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List Number 49809 dated 15.04.2013 or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/06/2015</td>
<td>Active</td>
<td>10044868</td>
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<td>STC Holder</td>
<td>STC Ref.</td>
<td>Rev.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>AIRCRAFT CABIN MAINTENANCE GmbH</td>
<td>ST01966CH</td>
<td>05CA-007</td>
<td>BD-700-1A-10, BD-700-1A-11</td>
<td>Installation of avionics equipment according to Virgin (V02) fleet standards</td>
<td>GOODRICH Master Data List (MDL) 2002-167 Rev C Dated August 19th, 2002 GD/GOODRICH Supplemental Instructions for Continued Airworthiness (ICAI) D02002-005 dated July 01st, 2002 GOODRICH Certification Plan CT73-00-0340-CP-01 dated April 20th, 2013</td>
<td>A copy of this certificate and GOODRICH NDL 2002-167 rev C dated August19th, 2002 (or later approved version) must be maintained as part of the once per mission for the modified aircraft.</td>
<td>28/10/2013</td>
<td>Active</td>
<td>10044903</td>
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<tr>
<td>LEARJET Inc.</td>
<td>ST00970SE-D</td>
<td>05AFM-017</td>
<td>G-1159, G-1159A, G-1159B</td>
<td>Installation of passenger seats</td>
<td>TASS Inc. Certification Plan</td>
<td></td>
<td>15/05/2013</td>
<td>Active</td>
<td>10044888</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>ST00710SE-D</td>
<td>ST01966CH</td>
<td>G-1159, G-1159A, G-1159B</td>
<td>Installation of provisions for dual CMC Electronic Flight Bag (EFB) system</td>
<td>FAA STC No. ST0208LA dated 17 December 2012 Learjet Inc. Master Data List (MDL) BAS0081085, Revision NC, dated 17th December 2012 or later revisions of the above listed documents approved by EASA</td>
<td>This modification applies to airplanes equipped with the Honeywell/ACCTCAS 2000 system only. The incorporation of Aviation Communication &amp; Surveillance Systems (ACSS) Service Bulletin SB08239-001 (FAA-A02012-02-06 relates), in its entirety with no deviations, in Revision 0 dated 26 June 2011, is a co.</td>
<td>15/05/2013</td>
<td>Active</td>
<td>10044895</td>
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<tr>
<td>TASS Inc.</td>
<td>ST001000SE-D</td>
<td>ST01966CH</td>
<td>G-1159, G-1159A, G-1159B</td>
<td>Installation of provisions for dual CMC Electronic Flight Bag (EFB) system</td>
<td>FAA STC No. ST0208LA dated 17 December 2012 Learjet Inc. Master Data List (MDL) BAS0081085, Revision NC, dated 17th December 2012 or later revisions of the above listed documents approved by EASA</td>
<td>This modification applies to airplanes equipped with the Honeywell/ACCTCAS 2000 system only. The incorporation of Aviation Communication &amp; Surveillance Systems (ACSS) Service Bulletin SB08239-001 (FAA-A02012-02-06 relates), in its entirety with no deviations, in Revision 0 dated 26 June 2011, is a co.</td>
<td>15/05/2013</td>
<td>Active</td>
<td>10044903</td>
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<tr>
<td>PENTASTAR AVIATION, LLC.</td>
<td>ST00710SE-D</td>
<td>ST01966CH</td>
<td>G-1159, G-1159A, G-1159B</td>
<td>Installation of provisions for dual CMC Electronic Flight Bag (EFB) system</td>
<td>FAA STC No. ST0208LA dated 17 December 2012 Learjet Inc. Master Data List (MDL) BAS0081085, Revision NC, dated 17th December 2012 or later revisions of the above listed documents approved by EASA</td>
<td>This modification applies to airplanes equipped with the Honeywell/ACCTCAS 2000 system only. The incorporation of Aviation Communication &amp; Surveillance Systems (ACSS) Service Bulletin SB08239-001 (FAA-A02012-02-06 relates), in its entirety with no deviations, in Revision 0 dated 26 June 2011, is a co.</td>
<td>16/05/2013</td>
<td>Active</td>
<td>10044906</td>
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</tbody>
</table>

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.
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<tr>
<th>STC Holder</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>Avionica Inc. SatLINK</td>
<td>10044932</td>
<td>Antenna in accordance with Ancohnine Holdings LLC.</td>
<td>Report STC 004/03/2013 for Continued Approval</td>
<td>- Reinstatement of an Avionica Inc. SatLINK communication management system with cabin wireless capability in Boeing 737-300-300-400-700-400 series aircraft and remain applicable to this certificate/approval. Prior to installation of this design change it must be determined that there is no adverse effect or repair will introduce noise or effect upon the airworthiness of the product.</td>
<td>17/05/2013</td>
<td>Active</td>
<td>10044932</td>
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<tr>
<td>STC TWENTY ONE LIMITED</td>
<td>10044926</td>
<td>Installation of SatLINK antenna in accordance with Ancohnine Holdings LLC.</td>
<td>Report STC 004/03/2013 for Continued Approval</td>
<td>- Reinstatement of an Avionica Inc. SatLINK communication management system with cabin wireless capability in Boeing 737-300-300-400-700-400 series aircraft and remain applicable to this certificate/approval. Prior to installation of this design change it must be determined that there is no adverse effect or repair will introduce noise or effect upon the airworthiness of the product.</td>
<td>21/05/2013</td>
<td>Active</td>
<td>10044926</td>
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<tr>
<td>AIR CRUISERS COMPANY</td>
<td>10044929</td>
<td>Installation of Air Cruisers SatLINK antenna in accordance with Ancohnine Holdings LLC.</td>
<td>Report STC 004/03/2013 for Continued Approval</td>
<td>- Reinstatement of an Avionica Inc. SatLINK communication management system with cabin wireless capability in Boeing 737-300-300-400-700-400 series aircraft and remain applicable to this certificate/approval. Prior to installation of this design change it must be determined that there is no adverse effect or repair will introduce noise or effect upon the airworthiness of the product.</td>
<td>21/05/2013</td>
<td>Active</td>
<td>10044929</td>
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<tr>
<td>STC 1500 Autopilot</td>
<td>10044930</td>
<td>Installation of SatLINK antenna in accordance with Ancohnine Holdings LLC.</td>
<td>Report STC 004/03/2013 for Continued Approval</td>
<td>- Reinstatement of an Avionica Inc. SatLINK communication management system with cabin wireless capability in Boeing 737-300-300-400-700-400 series aircraft and remain applicable to this certificate/approval. Prior to installation of this design change it must be determined that there is no adverse effect or repair will introduce noise or effect upon the airworthiness of the product.</td>
<td>21/05/2013</td>
<td>Active</td>
<td>10044930</td>
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<tr>
<td>STC 1500 Autopilot</td>
<td>10044932</td>
<td>Installation of SatLINK antenna in accordance with Ancohnine Holdings LLC.</td>
<td>Report STC 004/03/2013 for Continued Approval</td>
<td>- Reinstatement of an Avionica Inc. SatLINK communication management system with cabin wireless capability in Boeing 737-300-300-400-700-400 series aircraft and remain applicable to this certificate/approval. Prior to installation of this design change it must be determined that there is no adverse effect or repair will introduce noise or effect upon the airworthiness of the product.</td>
<td>21/05/2013</td>
<td>Active</td>
<td>10044932</td>
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<tr>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>Installation of ADEL7 - CPI S/G-16 Crash Position Indicator System</td>
<td>Master Data List document no. 2000-01-A-2103-R02, dated 25 April 2013 for Continued Airworthiness, document no. 2000-01-A-2211-R01 dated 06 April 2012</td>
<td>- Modification limited to A320N3 with Serial Numbers 9N6687 and 9N6690. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/05/2013</td>
<td>Active</td>
<td>10044932</td>
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<tr>
<td>J &amp; C ENGINEERING SERVICES Ltd.</td>
<td>Installation of a protruding camera on an approved fully provisioned installation</td>
<td>Certification Basis for the original product as amended therein or otherwise configured for installation in the Beechcraft King Air Model Airplanes/Replacement of the existing aircraft main batteries with a Concorde RG-380EK or RG-380EL series Lead Acid Battery.</td>
<td>As per flight manual supplement JCFAB3125 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/05/2013</td>
<td>Active</td>
<td>10044939</td>
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<td>CONCORDE BATTERY CORPORATION</td>
<td>Installation of the Concorde RG-380EK and RG-380EL series/RS/Abatieres in the Beechcraft King Air Model Airplanes/Replacement of the existing aircraft main batteries with a Concorde RG-380EK or RG-380EL series Lead Acid Battery.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/05/2013</td>
<td>Active</td>
<td>10044950</td>
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<tr>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>Installation of a Radar Attimeter Antenna Mount Kit. Rev. 1; Addition of model 206B.</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional later amendments if indicated on FAA STC. There are no changes to requirements for the associated certificated noise and/or emissions levels of the original product.</td>
<td>Only the installation of the Mount is approved. Installation of any of the provisions for which these STCs are approved is not allowed.</td>
<td>19/10/2018</td>
<td>Active</td>
<td>10044970</td>
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<tr>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>Retrofit Installation of a Dual CMC Electronics Class 3 Electronics Flight Bag (EFB) System.</td>
<td>Certification Basis for the original product and the EWIS CRI H-01 at issue 3 dated 20 August 2012 (v.2).</td>
<td>- The aircraft is dedicated to installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/05/2013</td>
<td>Active</td>
<td>10044973</td>
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<tr>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>Cargo Door Retention Kit</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>S/N 36191 and subsequent</td>
<td>24/05/2013</td>
<td>Active</td>
<td>10044994</td>
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### List of EASA Supplemental Type Certificates

**STC Holder**

**Certification basis**

**Date of Issue**

**Status**

**System Record**

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<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST03916AT</td>
<td>A321-212 / -231 / -232 / -241 / -271 / -281</td>
<td>Installation of the TEAC/Goodrich Digital Audio Reproducer and the TEAC/Goodrich Digital Video Reproducer for the Passenger Entertainment System (PES); Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification of noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA STC ST03916AT; Issued May 6, 2011; Project Specific Validation Report 2-95307-907, Rev. IR; dated Mar. 4, 2013; DL-2-95307-907, Rev. A; dated Apr. 28, 2011; Instruction for Continued Airworthiness 2-95307-805, Rev. A; dated Mar. 31, 2011; EASA/I.C.A. 2-95307-903, Rev. IR; dated Nov. 30, 2010; for later revision...</td>
<td>None</td>
<td>02/03/2016</td>
<td>Active</td>
<td>1004902</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>FALCON 2000, 2000EX</td>
<td>THRANE &amp; THRANE AVIATOR 350 SATellite COMMUNICATION SYSTEM WITH WIRELESS LOCAL AREA NETWORK Description as per Associated Technical Documentation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification of noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Documents List in document ref. 2081137 DOO issue 2, dated 24/05/2013; Master Drawings List in document ref. 2081137-GL issue 1, dated 27/03/2013; Certificates on Plan ref. 2081137-EO issue 2, dated 20/05/2013; Flight Manual Supplement ref. 2081137-AM issue 2, dated 23/05/2013; Maintenance Manual Sup...</td>
<td>None</td>
<td>24/05/2013</td>
<td>Active</td>
<td>10045018</td>
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<tr>
<td>AVIONIK STRAUSS ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td>AA-5, AA-5A, AA-5B, AG-58</td>
<td>Installation of Garmin G500 System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 3 (not required by 21.A.101) 29,251,303,305,561,575,603,605, 607,609,611,619,671,771,773,777, 803,867,954,982,1301,1303,130, 1307,1311,1327,1322,1323,1325,1326, 1327,1328,711...</td>
<td>- Flight Manual Supplement G500 ASR-2013-004- FMS-05.1-00 Rev. 00; Flight Manual Supplement GDL88 ASR-2013-004- FMS-05.2-00 Rev. 00; later revisions of the above listed documents approved by EASA...</td>
<td>A suitable Garmin GPS System must have been already installed. This STC includes the installation of Garmin’s Synthetic Vision (SVT) Option and the optional installation of Data Link Sensor System Garmin GDL88 antenna and Avionik Strauss Alti...</td>
<td>31/05/2013</td>
<td>Active</td>
<td>10045024</td>
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<tr>
<td>TURK HAVA YOLLARI TEKINM A.S TURKISH TECHNIC</td>
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<td>A321-231</td>
<td>CONVERSION OF CABIN LAYOUT FROM 210 Pax Full EIC TO 194 Pax Class BAND MOVABLE CLASS DIVIDER (MCD) INSTALLATION.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification of noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Design Declaration CG-25-0005-DD01-01; dated 20.04.2013; or later revisions of the above listed documents approved by EASA...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise and/or emissions that will impact the airworthiness of the product. The installation of this modification by third p...</td>
<td>27/06/2013</td>
<td>Active</td>
<td>10045055</td>
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<td>BOMBARDIER AEROSPACE</td>
<td></td>
<td>TCCA SA04-102 ISSUE 1</td>
<td>BD-100-1A10</td>
<td>Activation, Rockwell Collins 5000/6000 SATCOM</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification of noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Modification Data Summary Number E026000 Rev. 1; dated 20130411; or later revisions of the above listed documents approved by EASA...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise and/or emissions that will impact the airworthiness of the product...</td>
<td>28/05/2013</td>
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<td>10045060</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10045063</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>A109S</td>
<td></td>
<td>A109S installation of EGPWS MK XXII</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: CS-27 Amdt. 2, paragraph 25, 29(a)(r) 301, 303(c)(v) 305, 307, 361, 601, 603, 605 607, 608, 610, 611, 613, 619, 625, 771...</td>
<td>RFMS for installation of Honeywell MK XXII EGPRWS, Document No. 2810794078-0061, revision 2; dated 01.03.2013.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other subsequently installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/05/2013</td>
<td>Active</td>
<td>10045063</td>
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<tr>
<td>10045079</td>
<td>BLR AEROSPACE, L.L.C</td>
<td>REV. 4</td>
<td>FAA STC SR095055RC-D</td>
<td>AB 412, AB 412 EP 412, 412EP</td>
<td>Installation of FastFin® System Installation of a Dual Tailboom Strakes and FastFin® System Revision 1 mirrors the amendment of FAA STC SR095055RC-D. An approvedmoderated list was added and Rotorcraft Flight Manual Supplements were revised to include CABIN Numbers that incorporate Bell Helicopter Techins.</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: FAR91.629 Amdt. 29-40, FAR 29.2571 Amdt. 29-20; Appendix A29.4 Amtd. 29-2F. The requirements for environmental protection and the associatedcertified noise and/or emissions...</td>
<td>BLR RPM Supplement RFMS-412FT-F, Rev. IR (applicable to AB412 equipped with PT6T-5B, operating to Agusta Flight Manual AB412(R) RPM Supplement RFMS-412FT-F, Rev. IR (applicable to AB412 equipped with PT6T-3BE, operating to Agusta Flight Manual AB412(R) RPM Supplement RFMS-412FT-F, Rev. IR (applicable...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other subsequently installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/05/2019</td>
<td>Active</td>
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<td>10045086</td>
<td>GFD GmbH</td>
<td>REV. 7</td>
<td>LEARJET 31A/25A/36A 31A, 35A, 36A</td>
<td>SM GPS Antenna ProvisionRev. 1: Transfer to GFD GmbH</td>
<td>The Type and OSI Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>- ICA Instruction for Continued Airworthiness Data List Doc. No. 5253-0606, Master Data List Doc. No. 5253-0600; later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other subsequently installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
<td>Active</td>
<td>10045086</td>
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<td>10045092</td>
<td>JCB AERO 5A</td>
<td>A335-222</td>
<td></td>
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<td>Staircase Removal Modifications Description - Summary: Some Cabin items removed (Galley G7, Lavatory L7, VCC, Staircase, Wardrobe, Lavatory L3 Modified &amp; relocated,provisions for LDMCR (which is not installed). For further details according to Certification Plan A332-00324-01-CP-01...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>Project Data List A332-00324-01-PDL-01, rev CCertification Plan A332-00324-01-CP-01, issue DICA, A332-00324-01-DICA-01 Rev (Ror later revisions of the above listed document approved by EASA).</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other subsequently installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/05/2013</td>
<td>Active</td>
</tr>
<tr>
<td>10045119</td>
<td>JET AVIONICS S.R.L</td>
<td>PA-31T CHEYENNE II</td>
<td></td>
<td>General avionics Improvements/ Installation/Replacement of COMNAV4/A GPS, TAWS/FMS, DME, Audio Panel, Multifunctional Display</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: CS-23.365, 561, 573, 587, 867, 1309, 1311, 1351, 1431, 1529 at their initial amendment/The requirements for environmental protection and the associatedcertified noise and/or emissions...</td>
<td>-Master Drawing List BD 003-10 Rev. 1 11.03.2013, Maintenance Manual Supplement BD 003-10-14 Rev. 2 11.03.2013, Flight Manual Supplement GNS-530, TS400, Flightmax FSBD BD 003-10-16Rev. 0 30.11.2012,-2Rev. 1 04.02.2013, -3 (Rev. 0 30.11.2012)-MME. Update BDI 003-10-17 Rev. 1 04.02.2013 or later r...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other subsequently installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/05/2013</td>
<td>Active</td>
<td>10045119</td>
</tr>
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</table>

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this design change it must be determined that the interrelationship between the design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Installation of Wescam MX-10 on Airbus A330 and A340 aircraft. This STC replaces the original logo lights by LED logo lights.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this design change it must be determined that the interrelationship between the design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.
<table>
<thead>
<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>10045202</td>
<td>A&amp;D AEROSPACE ARWORTHINESS</td>
<td>REV. 2</td>
<td></td>
<td>A330-341 / -342 / -343</td>
<td>Activation of GSMConnect SystemGSMConnect is a solution that provides GSM and WLAN services to aircraft passengers. It consists of hardware and software parts for both in-craft and the ground, with the airborne GSM and WLAN network being connected to ground network via satellite.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25.425-29 / 25.581 / 25.771(a)(c) / 791(c)(d) / 896(a) / 896 / 1301 / 1307(d) / 1309(a)(b)(c) / 1310 / 1320-2.</td>
<td>None</td>
<td>20/06/2014</td>
<td>Active</td>
<td>10403102</td>
<td></td>
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<tr>
<td>10045229</td>
<td>ROEDER PRECISION GMBH</td>
<td>REV. 1</td>
<td></td>
<td>PA-28-160 ARROW</td>
<td>Installation of J.P. Instruments (JPI) Primary Engine Data Management System EDM930 in accordance with Engineering OrderEADM930- NVY32818-05. The EDM-930 is a combined electronic indicator system which simultaneously displays to the pilot, powerplant and aircraft systems operating parameters.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23, Amendment 2 for affected paragraphs 23.21, 23.303, 23.561, 23.803, 23.137, 23.1309, 23.1321, 23.1327, 23.1337, 23.1357, 23.1365, 23.1431, 23.1520...</td>
<td>None</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10405229</td>
<td></td>
</tr>
<tr>
<td>10045232</td>
<td>AVIONICARE LIMITED</td>
<td></td>
<td>727-100</td>
<td>Upgrade of Dual Trimmer HT-9100 GNSS to Dual Universal UNS-1vE FMS on Boeing 727 Aircraft.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI-H-01 'Enhanced Airworthiness Programme for Airplane Systems - ICA on EIVS', the following paragraph(s) at a later amendment: CS 25 Amendment 4.C...</td>
<td>Avionicare Master Document List/Avionics Catalogue AMD-004 Issue 01 for the above listed documents approved by EASA.</td>
<td>Applicable to sn 18934 only. Prior to installation of this design change it must be determined that there will be no adverse effect on the airworthiness of the product.</td>
<td>10/06/2013</td>
<td>Active</td>
<td>10405232</td>
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<tr>
<td>10045252</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC 503055AT</td>
<td>MD900</td>
<td>Installation of an Concorde replacement lead acid battery for MD900Series</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certified/approved FAR Part 27 (as defined in FAA TCOG/H1WMM, 37.1301(a), (b), (c), (d), 1309(a), (b), (c), (d) &amp; (e), 1529. The requirement environmental protection and the associated criteria not...</td>
<td>Prior to installation of this design change it must be determined that there will be no adverse effect on the airworthiness of the product.</td>
<td>21/06/2013</td>
<td>Active</td>
<td>10405252</td>
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</tbody>
</table>
**List of EASA Supplemental Type Certificates**

### 10045274

**STC Holder**: LUITTHANSA TECHNIX AG  
**STC No.**: A330-300  
**Description**: DUL A330 - Italian Style Interiors - Summary - Installation of new Y-Class, C-Class and First Class seats and lining.  
**Certification Basis**: For the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/modified noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.  
**Associated Technical Documentation**: - Certification Plan CP5-12/116 Iss 1 at rev 2. Supplemental Type Design Definition Document STDDCS-12/116 Iss 1 rev 0 (Master document and definition). ICA, ref: A-00-1240-ICA, rev 01 - Compliance Record List CRLS-12/116 Iss 1 at rev 0 later revisions of the above listed documents approved by EASA.  
**Limitations/conditions**: Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.  
**Date of Issue**: 12/06/2013  
**Status**: Active  
**System Record**: 10104274

### 10045275

**STC Holder**: SUPERJET INTERNATIONAL S.P.A.  
**STC No.**: RRU-95B  
**Description**: SSJ-100 - Italian style Interiors - Installation Original STC is installing Italian Style interiors in SUHO1 RRU-95S (Aircraft). Change Notice CN-UJ93-0022 introduces Lavatories reinforcement and removes the AFM Limitations. Change Notice CN-UJ93-0025 introduces structural reinforcement on gallery22 to improve flight deck safety.  
**Certification Basis**: For the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/modified noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.  
**Limitations/conditions**: Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.  
**Date of Issue**: 08/03/2017  
**Status**: Active  
**System Record**: 10045275

### 10045282

**STC Holder**: DUNLOP AIRCRAFT TYRES LIMITED  
**STC No.**: CAA UK RESPONSIBILITY ATR 72-211/-212/-212A ATR 72-101/102-201-202  
**Description**: Installation of Dunlop Aircraft Tyres on ATR 72-212 model. The STC installs Dunlop Aircraft Tyres Ltd PN D220730T (Main LandingGear) and PN DR155007 (Nose Landing Gear).  
**Certification Basis**: For the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/modified noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.  
**Associated Technical Documentation**: Master Document List, ref. DALT_Pr MDL Rev IR, dated 20 May 2013 Instruction for Continued Airworthiness, ref. DALT_PRICHA Rev IR, dated 20 May 2013 or later revisions of the above listed documents approved by EASA.  
**Limitations/conditions**: None.  
**Date of Issue**: 12/06/2013  
**Status**: Active  
**System Record**: 10045282

### 10045283

**STC Holder**: CHALLENGER AVIATION PRODUCTS INC.  
**STC No.**: FAA STC SEAL2352CH SEE EASA APPROVED MODEL LIST  
**Description**: Direct Oil Filter Replacement - EASA Validation of FAA STC SEAL2352CH (Installation of Challenger LeakFree oil filter assembly in accordance with Challenger Aviation Products, Inc. Installation Instructions 1124 Revision B, dated October 31, 2009, or later FAA approved revision).  
**Certification Basis**: For the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/modified noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.  
**Associated Technical Documentation**: Challenger Aviation Products, Inc. Installation Instructions 1124 revision B, dated October 31, 2009 or later revisions of the above listed documents approved by EASA.  
**Limitations/conditions**: 1. Compatibility of this design change with previous modifications must be determined by the installer. 2. Use of Challenger Aviation Products, Inc. Inspection and Cleaning Instructions 1125, Revision C, dated October 31, 2009 or later FAA approved revisions is required. 3. A copy of this certificate must be affixed to the aircraft.  
**Date of Issue**: 12/06/2013  
**Status**: Active  
**System Record**: 10045283

### 10045284

**STC Holder**: TATENHILL AVIATION LTD.  
**STC No.**: CAAC UK RESPONSIBILITY PA-31  
**Description**: Approval of an Aspen EFD-1000 Primary Flight Displays against the requirements of EU-OPS for Dual Pilot Operations.  
**Certification Basis**: For the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/modified noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.  
**Associated Technical Documentation**: Master Document List/MDL. 00812 Revision B Installation Manual/ATM. 00812 Revision A Installation Instructions for Continued Airworthiness ICA. 00812 Revision AFFIG FT Document Supplement/OPS. 00812 Revision Ari later revisions of the above listed documents approved by EASA.  
**Limitations/conditions**: None.  
**Date of Issue**: 12/06/2013  
**Status**: Active  
**System Record**: 10045284

### 10045290

**STC Holder**: UUDS  
**STC No.**: A330-200  
**Description**: Contour A330-200 Business Class seat modification and bumper installation. This modification removes the existing RockwellCollins IFE equipment monitors and harnesses from Contour Business class double seats Model/200300-001 and replaces existing Amsafe seat belts. The seats are modified to provide...  
**Certification Basis**: For the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/modified noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.  
**Associated Technical Documentation**: Master Modification Sheet MM- TMS99-065-25-001-00r later revisions of the above listed documents approved by EASA.  
**Limitations/conditions**: Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.  
**Date of Issue**: 13/06/2013  
**Status**: Active  
**System Record**: 10045290
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<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10045296</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 6</td>
<td>FAA ST04520AT-D</td>
<td>GVI (G600), GVI (G600ER), G1n (G650)</td>
<td>SATCOM upgrading, IFE/CMS, WIFI and 10045310</td>
<td>ST0420ST-D, Installation of airframe Interior. Date Amended 19 OCT 2021</td>
<td>2021AFM Supplement ref. GC64184M001 rev. 2 on 2014/2021/AASA Validated Configuration Lists. GC6204002, 2021/2022/AASA Validated Configuration Lists.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/11/2021</td>
<td>Active</td>
<td>10045296</td>
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<tr>
<td>10045303</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
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<td></td>
<td></td>
<td>Installation of HATS Tracking System, Limitations/Conditions</td>
<td></td>
<td></td>
<td></td>
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<td>14/02/2013</td>
<td>Active</td>
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<tr>
<td>10045310</td>
<td>AHONIK STRAUBUNG ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>Single Installation GPS/COMM/NAV System Garmin GTN650/660/760/770</td>
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<td>14/03/2013</td>
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<tr>
<td>10045316</td>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>Revision 1 of STC 10045316, addition of models to models 200-200C/2000/2017</td>
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<td></td>
<td>12/03/2018</td>
<td>Active</td>
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<tr>
<td>10045334</td>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td></td>
<td></td>
<td></td>
<td>Upgrade to CPDLC for ATN B1 capability upgrade of existing Communication Management Unit software for CPDLC (priorise the Rockwell-Collins solution) with an ATN B1compliant version.</td>
<td></td>
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<td></td>
<td>17/06/2013</td>
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<tr>
<td>10045336</td>
<td>SABENA TECHNICS BOD</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>SATCOM upgrading, IFE/COMS, WIFI and G1M new systems modification</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td>18/03/2013</td>
<td>Active</td>
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<td>STC Holder</td>
<td>STC Ref.</td>
<td>Rev. level</td>
<td>Foreign/NAA Ref.</td>
<td>STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>KUERZI AVIONICS AG</td>
<td>10045344</td>
<td>FALCON TX</td>
<td>GMS &amp; WI-Fi Integration/New installed equipment - TriaNicolsys, 21TR10-XU-07-00 AEROBOT75+ (GMS Base Transceiver Station(RTS)), TriaNicolsys, 21TRB0-00-01-00 NCU (Network Control Unit), VT Mispw, 90983-1-24 AP (Cabin Network Router), Mobile Mark, RMM-UMB-Q4-4B, GSM/WIFI TX Antenna, Mobile Mark, RMM-UMB-Q4...</td>
<td></td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise/level or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List Doc. No. MDL-2011016599, Rev. A, 08.06.2013; FMS7 Doc. No. FMS-2011016599, Rev. 0, 28.05.2013; ICA Doc. No. ICA-2011016599. Rev. N/C, 29.05.2013 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>19/06/2013</td>
<td>Active</td>
<td>10045344</td>
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<tr>
<td>A.S. AVIONICS SERVICES S.A.</td>
<td>10045384</td>
<td>EMB-135BJ</td>
<td>Installation of HWG-A4 Potable Water Treatment Unit on EMB-135BJ/EASA validation of ANAC STC 2012/009-08</td>
<td></td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td></td>
<td>Applicable to EMB-135BJ S/N 14500979, 14501029 and on/for installation of this design change it must be determined that the interrelationship between this design change and anything previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/06/2013</td>
<td>Active</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>10045378</td>
<td>REV. 3 FAA ST110415C</td>
<td>MF50, F50EX</td>
<td>Installation of upgraded hardware to replace Rockwell Collins Proline 400 displays with Rockwell Collins Pro Line 21 displays, add an Integrated Flight Information System (IFIS) and enable localizer performance with vertical guidance (LPV) capability Rev. 3. Transfer from BHE &amp; Associated, Ltd. to Dassault...</td>
<td></td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Drawing List ref. EMB135BJ-38-0001MD, Rev. D, dated 11/09/2012 Certification Plan ref. EMB135BJ-38-001CP, Rev.B, dated 07/03/2012, supplemented by EASA Declaration of Compliance ref.EMB135BJ-38-0001EA-SA-DC, Rev. Original, dated 20/02/2013 Installation Manual ref. EMB135BJ-38-0001M, Rev. C.</td>
<td>The installation requires the installation of EASA STC 10064901 (FAA STCST01833WI) and/or EASA STC 10061518 (FAA STC ST01433WI-D) in order to provide global positioning system datathat is synchronized with the wide area augmentation system. For airworthiness of the product, this approval is limited to Master Data List N20020301 Issue &quot;001&quot; or later FAA approved revision.</td>
<td>19/03/2020</td>
<td>Active</td>
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<tr>
<td>STUDENT DEVELOPMENT Inc.</td>
<td>10045381</td>
<td>REV. 2 FAA ST02436LA</td>
<td>737-700/-800/-900ER</td>
<td>Installation of Cabin Interior Noise Reduction System.</td>
<td></td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>1. N2 Development, Inc. Master Data List N2002 Rev. D dated 08 October 2012 or later FAA approved revision 2. N2 Development, Inc. Instructions for Continued Airworthiness Documentation N20020361 Issue &quot;01&quot; dated 28 April 2012 or later FAA approved revision.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. 2. This approval is limited to Master Data List N2002 Issue &quot;01&quot; or later FAA approved revision.</td>
<td>18/09/2018</td>
<td>Active</td>
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<tr>
<td>ZIM AIRCRAFT CABIN SOLUTIONS, LLC</td>
<td>10045384</td>
<td>FAA STC ST01735AT-D</td>
<td>MD-11</td>
<td>VHF Installation - MD-11 Collins VHF Transceivers and GateWay Radios Tuning Panels installation toenable 8,33 kHz wide area augmentation system</td>
<td></td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>TIMCO Master Data List No. 997058L001 Rev. B dated 29 October 1995 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions off this decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/06/2020</td>
<td>Active</td>
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<tr>
<td>AIM AVIATION (HENSHALLS) LTD</td>
<td>10045392</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>A330-200</td>
<td>Installation of Galleys and associated ceiling panels into an Airbus A330-200 to be operated by Delta.</td>
<td></td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>AIM Aviation Ltd Definition Document No. LS1240 Issue New Dated 15 May 2013 or later revisions of the above listed documents approved by EASA...</td>
<td>None</td>
<td>20/06/2013</td>
<td>Surrrendered</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10045396</td>
<td>SABENA TECHNICS BGC</td>
<td>REV 4</td>
<td></td>
<td>A321-231/-233/-232</td>
<td>The Certification Basis for the original product as amendability the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: Applicable regulations at CS 25 Amendment 4 (refer to associate Certification Plan 0594-01-A-2101-R05, section 5, for details)</td>
<td>Instructions for Continued Airworthiness, ref : 0594-01-A-2211-R05for later revisions of the above listed documents approved by EASA-Master Data List ref : 0594-01-A-2103-R05</td>
<td>As per design of this STC, the modifications related to TCAS 7.0 need to be previously installed by an Airbus modification (refer to Aircraft System Configuration Checklist reference 0594-01-A-ASSC-R06) and shall correspond to the description included in System Technical Specification reference 0594-01...</td>
<td>14/02/2018</td>
<td>Active</td>
<td>10045396</td>
<td></td>
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<tr>
<td>10045398</td>
<td>HELIFAB Inc.d/b/a ARROW AVIATION CO. LLC</td>
<td></td>
<td></td>
<td>BELL 212, 412, 412EP</td>
<td>Installation of Cabin Area Cargo Net, according to Master Drawing 8Documents List HF-412-DL-101, Rev. L, in Bell 212, 412, 412EPHelicopters</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unarunchanged and remain applicable to this certificate approval</td>
<td>JHF-412-DL-101, Rev. L, Master Drawing &amp; Documents List HF-412-CCL-0003, Rev. R, Compliance Checklist, HF-412-ST-0002, Rev. C, Load Test Plan (only Load Test results, Table 1); HF-413-L2550-01, Rev. E, Cargo Net Installation (Drawings)</td>
<td>Prior to installation of this design change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>24/06/2013</td>
<td>Active</td>
<td>10045424</td>
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<tr>
<td>10045413</td>
<td>E A D AEROSPACE AIRWORTHINESS TIAECLIPSE TECHNICS</td>
<td></td>
<td></td>
<td>A320-231/-232/-233</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval</td>
<td>Certification Program 01381-EAS-CPM-01 Rev C, dated 01/03/2013, Compliance Checklist 01381-EAS-A-CCL-01 Rev A, dated 12/06/2013, Master Drawing List 01381-EAS-A-MDL Rev C, dated 12/06/2013, Aircraft Flight Manual Supplement 01381-EAS-A-MLST Addendum 01 Rev. A, dated 09/06/2013FCC Instruction a for Continued Airworthiness</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>21/06/2013</td>
<td>Active</td>
<td>10045413</td>
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<tr>
<td>10045423</td>
<td>Bristow Helicopters Ltd</td>
<td></td>
<td></td>
<td>EC 155 B1</td>
<td>Certification Basis for the original product as amendability the following additional or alternative airworthiness requirements: CS 25 Amendment 3/CS 29.1301/CS 29.1290/CS 29.1431/CS 29.1581The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unarunchanged and remain applicable to this certificate approval</td>
<td>Bristow Helicopters Limited Eurocopter EC155 Flight Manual Supplement No. 1 Modification MOD/155/34/110/05 Final Issue dated 22 August 2013, Aircraft Flight Manual Supplement 01381-EAS-A-MLST Addendum 01 Rev. A, dated 09/06/2013FCC Instruction a for Continued Airworthiness</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>24/06/2013</td>
<td>Active</td>
<td>10045423</td>
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**List of EASA Supplemental Type Certificates**

<table>
<thead>
<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tbody>
<tr>
<td>10045424</td>
<td>HELIFAB Inc.d/b/a ARROW AVIATION CO. LLC</td>
<td></td>
<td></td>
<td>BELL 212, 412, 412EP</td>
<td>Installation of Cabin Area Cargo Net, according to Master Drawing 8Documents List HF-412-DL-101, Rev. L, in Bell 212, 412, 412EPHelicopters</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unarunchanged and remain applicable to this certificate approval</td>
<td>JHF-412-DL-101, Rev. L, Master Drawing &amp; Documents List HF-412-CCL-0003, Rev. R, Compliance Checklist, HF-412-ST-0002, Rev. C, Load Test Plan (only Load Test results, Table 1); HF-413-L2550-01, Rev. E, Cargo Net Installation (Drawings)</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>24/06/2013</td>
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<td>STC Holder</td>
<td>Rev.</td>
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<td>Certification basis</td>
<td>Limitations/Conditions</td>
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<td>TELEDYNE UK LIMITED</td>
<td>REV 1</td>
<td>CAA UK</td>
<td>MD-11, MD-11F</td>
<td>Installation of Wireless Quick Access Recorder Rev 1: Transfer from Teledyne Limited to Teledyne UK Limited.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: JAR 25 Amendment 16 §25.23. §25.26. §25.27. §25.561. §25.581(b)(1). §25.603 (A)(b)(c). §25.605(b). §25.811. §25.869(a)(4)…</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. Compatibility has been established for the aircraft.</td>
<td>26/11/2020</td>
<td>Active</td>
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<tr>
<td>FR AVIATION Ltd./BA COBHAM AVIATION SERVICES UK</td>
<td>REV 2</td>
<td>CAA UK</td>
<td>DO22B-202K</td>
<td>Introduction of Oil Spill Response Role System Rev 1: 1-23The Certification Basis for the original product is applicable to this certificate/approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AFMS FR494FM0121 dated 06 Feb. 2014 for Installation of eXPhone dedicated equipment and activation of the in-flight GSM connectivity system.</td>
<td>The approval of this design change is conditional on the large belly/radome being installed. UK CAA Major Modification A20895 refers. SN: 1175 Prior to installation of this modification it must be determined that the relationship between this modification and any other previously installed modification is unchanged and will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2014</td>
<td>Active</td>
<td>10045438</td>
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<tr>
<td>BOMBARDIER AEROSPACE</td>
<td>C-LSA07-156/ D</td>
<td>Arcell ST3100 Satcom Installation</td>
<td>BD-700-1A10</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC No. C-LSA07-156/ D, Issue 2 dated 17 December 2008. Bombardier Aerospace Master Document List BAS67017018, Revision B dated 04 October 2008. Bombardier Aerospace Instructions for Continued Airworthiness Document BAS67070200, Revision A dated 24 October 2008 or later revisions of thea…</td>
<td>Restricted to ASR 9003/Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification is applicable to this product.</td>
<td>25/06/2013</td>
<td>Active</td>
<td>10045436</td>
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<tr>
<td>SABENA TECHNICS BOCG</td>
<td>777-300ER</td>
<td>sXPhone System activation on B777-300ER with VVIP cabin interior [Installation of sXPhone dedicated equipment and activation of their flight GSM connectivity system]</td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, Aerocore document No. 0616-01-A-2103, Revision 0 dated 14 June 2013. Aircraft Flight Manual Supplement, Aerocore document No. 0616-01- A-Z501-0-000000000, Revision 1 dated 11 June 2013. Instruction for Continued Airworthiness, Aerocore document No. 0616-01- A-2211, Revision 1.</td>
<td>Prior to installation of this modification it is limited to the following B777-300ER with VVIP cabin interior, except similarity demonstrated per aerocore document 0011-01-A-0914 (Classification of Modification/Change Procedure for GSM Co…</td>
<td>25/06/2013</td>
<td>Active</td>
<td>10045443</td>
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<tr>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>REV 1</td>
<td>PC-12, PC-1245</td>
<td>PC-12/47, PC-12/47E</td>
<td>Installation of L-3 Communications WX-500 Stormscope with Arantena NY-163 Flight Manual Supplement AER-2013-001-PM5-45-01 Amtd. 01/04/2017 or later revisions of the above listed documents approved by EASA.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 3, 23,301, 303, 305, 307, 365, 561, 571, 575, 603, 605,607, 609, 611, 613, 617, 771, 773, 843, 863, 867, 1301, 1309, 1311, 1312, 1327, 1351, 1357, 1359, 13,…</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/02/2017</td>
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<td>10045451</td>
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<td>EASA STC No.</td>
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<td>Certification basis</td>
<td>Associated Technical Document</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>10045456</td>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>REV. 2</td>
<td></td>
<td>777-300/-300ER</td>
<td>GSM and WiFi activation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Design List, EAD Aerospace document No. 01410-EAS-A-MDL. Revision F, dated 30 June 2017. Instructions for Continued Airworthiness, EAD Aerospace document No. 01410-EAS-A-ICA-01. Revision E, dated 16 June 2017. Aircraft Flight Manual Supplement, EAD Aerospace document No. 01410-EAS-A-ATM-01, Rev. E</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/07/2017</td>
<td>Active</td>
<td>10045456</td>
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<tr>
<td>10045472</td>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td></td>
<td></td>
<td>737-800(-900/-900ER 737-300/-400/-500/-600/-700)</td>
<td>SpectraLink+ w/CPDLC installation (Link2000+ IATN B1 compliant) in Boeing 737 aircraft</td>
<td>CS 25 Amendment 12 as amended by the following Special Conditions F-01 High Intensity Radiated Field (HIF) Data Link Services for the Single European Sky. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Design List, MDL282-050 Issue 04 Flight Manual Supplement FMS/282-002 Issue 01 (Maintenance Manual Supplement MM5/282-012 Issue 01 Service Bulletin EIRTECH-S-162 Issue 01) for later revisions of the above listed documents approved by EASA</td>
<td>This CPDLC installation is limited to providing a supplementary means of communication; voice shall remain the primary means of communication. CPDLC shall only be used for routine datalink exchanges during en-route operations in upper airspace. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/06/2013</td>
<td>Active</td>
<td>10045472</td>
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<tr>
<td>10045479</td>
<td>HAWKER BEECHCRAFT LTD.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>PREMIER 1/1A</td>
<td></td>
<td>To introduce an Aviator Swift Broadband SatCom System with wifi.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Hawker Beechcraft Ltd Modification/SC0485A. Operation in accordance with Hawker Beechcraft Ltd Flight Manual Supplement No 2 Issue 1 to Manual 390-5801-003. Maintain in accordance with Hawker Beechcraft Ltd ICA document MDL50C-C-028-A04368 Issue 1 dated June 2013.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/06/2013</td>
<td>Active</td>
<td>10045479</td>
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<tr>
<td>10045481</td>
<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A319-133</td>
<td>Galley Complex Installation.Installation of Galleys D1 and G2 and a Sliding Door.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CR1 D-01 Installation of Interior Doors. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are...</td>
<td>Modification TEN 722, Airbus A319-133, Galley Complex Installation Issue 1 dated 10 June 2013.</td>
<td>Airbus A319-133 serial number 4842 only; This Galley Complex Installation is part of another EASA STC for this aircraft. Interior Outfitting Installation, EASA Project Numbers0010014342-001 and EASA STC Approval 10046155 refer. This Galley Complex Installation must not be installed without the Interior O...</td>
<td>23/06/2013</td>
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<td>10045481</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10045486</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST02059NY-D</td>
<td>A330-341, A330-342, A330-343, A330-321, A330-322, A330-323, A330-301, A330-302, A330-303</td>
<td>Installation of Panasonic Global Communication System in A330</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/associated noise emission levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data Uit 0154-07357-2272 Rev. -</td>
<td>This Approval is linked to the approvals hereunder, which are out of the scope of this STC and must be covered by another approval. The demonstration of compatibility of the associated noise and emission levels of the original product as amended by EASA ED.:</td>
<td>28/06/2013</td>
<td>Active</td>
<td>10045486</td>
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<td>10045487</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH13-18</td>
<td>MBK BK117 A1A/A1B/B</td>
<td>D117-840 Beapaw Installation/Installation of Beapaw as defined in Master Document List MLD-D117-840</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/associated noise emission levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List MLD-D117-840 Rev. A, Installation Instructions IN-D117-840 Rev. A, Instructions for Continued Airworthiness ICA-D117-840 Rev. 0</td>
<td>None</td>
<td>28/06/2013</td>
<td>Active</td>
<td>10045487</td>
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<td>10045488</td>
<td>airbus engineering GmbH</td>
<td>REV. 3</td>
<td>208, 208B</td>
<td>Installation of Parachute Kit consisting of: Manual retractor/Manifold, hand/holdfast, cabin bench</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/associated noise emission levels of the p.</td>
<td>Refer to Technical Publication - Applicability Matrix (PC+/-). The revision of this STC may add the approved/approved/certificated list of noise and/ or emission changes.</td>
<td>All limitations defined in AFMS. Prior to installation of this modification all limits must be determined by EASA.</td>
<td>13/03/2023</td>
<td>Active</td>
<td>10045488</td>
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<tr>
<td>10045492</td>
<td>FR AVIATION Ltd/DIA COBHAM AVIATION SERVICES UK</td>
<td>CAA UK RESPONSIBILITY</td>
<td>DORNER 228-220K</td>
<td>Introduction of Flight Management System and RNP5 and GPS system with a new flight Management System/Providing an RNP5 navigation solution and an amended B-RNAV approval.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/associated noise emission levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AFMS FRA9HF M0122 Rev. 0 dated 14 Jun 2018, Build in accordance with Document List in DC-0233: Iss. 1, and later additions.</td>
<td>The installation of this modification by third persons is subject to written permission of the approval holder and disposal of the appropriate documentation.</td>
<td>28/06/2013</td>
<td>Active</td>
<td>10045492</td>
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<tr>
<td>10045512</td>
<td>CESSNA DUESSELDORF CITATIONSERVICES CENTER GmbH</td>
<td>REV. 3</td>
<td>525, 525A</td>
<td>P-RNAV Approval of UNS-1K</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/associated noise emission levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>None</td>
<td>01/07/2013</td>
<td>Active</td>
<td>10045512</td>
<td></td>
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<tr>
<td>10045515</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>REV. 1</td>
<td>A340-313</td>
<td>Interior reconfiguration of Cathay Pacific A340 Aircraft/Premium economy class seat installation. For additional information refer to NAT Certification Plan N0406-D01NB</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/associated noise emission levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification plan No406-D01 rev BM (Master Drawing List : NA4006-D01 rev IG Certifications Summary) and No406-D00 rev AICA ref No406-D01 rev IA EASA/FTS ICA: Document : N0406-D01 rev II later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modifications and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/07/2013</td>
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<td>10045515</td>
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<td>EASA STC no.</td>
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<td>FAA STC/STC ref</td>
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<td>Associated Technical Documentation</td>
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<td>10046562</td>
<td>TELAIR INTERNATIONAL Inc.</td>
<td>FAA STC ST02148A</td>
<td>767-300BCF</td>
<td>Installation of Cargo Handling System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/authorized noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Texar International Index List. ER169101, Revision A, 05/06/2013 later revisions of the above listed documents approved by EASA.</td>
<td>04/07/2013</td>
<td>Active</td>
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<tr>
<td>10045959</td>
<td>ASI INNOVATION</td>
<td>L-1329 (JETSTAR 731)</td>
<td>14 CPR Part 25, Amdt 136, IAW ASI-ICL-0136-01</td>
<td>Replacement of Flight Management System in accordance with Service Bulletin ASI-SB-L1329-3400-0135 Rev 2</td>
<td>Service Bulletin ASI-SB-L1329-3400-0135 Rev 2 2AFM Supplement ASI- AFM-0135-09AS-ASI- MPS-0135-07Technical File ASI-SDL-0135-02r later revisions of the above listed documents approved by EASA.</td>
<td>Approval Limited to Airplane ConvertedFreighter L-1329 MSN 50292Pr to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the pro...</td>
<td>05/07/2013</td>
<td>Active</td>
<td>10045959</td>
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<tr>
<td>10046562</td>
<td>EUROCOPTER CANADA Ltd.</td>
<td>TCCA STHC-20</td>
<td>EC130B4</td>
<td>Operation with the following cabin door configuration: LH pilot door closed, LH sliding door open and locked, RH doors/panel removed.IAS STC is the validation of TCCA STC SH10-20 (Issue date April 22,2010)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/authorized noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Eurocopter Canada Limited Flight Manual Supplement FMS-ECL-137 rev. 0or later revisions of the above listed documents approved by EASA.</td>
<td>12/07/2013</td>
<td>Active</td>
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<tr>
<td>10046568</td>
<td>AIRCRAFT CABIN MAINTENANCE GmbH</td>
<td>737-900, 737-900ER, 737-800, 737-700, 737-700, 737-300, 737-400, 737-800, 737-100, 737-200, 737-200C.</td>
<td>Installation of refurbished Passenger Seat Belt P/N’s 27741-2-01-40 and 27741-2-01-60 on 16g seats</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/authorized noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Refer to Master Document List E/0 No. 126-1003837 Rev. 0 (21.06.2013)for later revisions of the above listed documents approved by EASA.</td>
<td>This approval of operation with cabin doors / panel removed does not constitute approval of installation of those doors / panel being removed. Installation of all doors / panels being removed in order to obtain the configuration covered by this approval shall be covered by dedicated EASA approvals.Prior...</td>
<td>10/07/2013</td>
<td>Active</td>
<td>10046568</td>
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<td>10046665</td>
<td>STANDARD AERO</td>
<td>FAA STC STO3225CH</td>
<td>ERJ 170-200 STD ERJ 170-200 LR ERJ 170-100 STD ERJ 170-100 LR</td>
<td>Installation of a portable emergency locator transmitter (ELT) and/or modification of interior placards.EASA validation of Basic FAA STC ST03225CH.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/authorized noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Standard Aero Master Data List 1024545. Revision (C), datedJune 25,2013for later revisions of the above listed documents approved by EASA.</td>
<td>10/07/2013</td>
<td>Active</td>
<td>10046665</td>
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<tr>
<td>10046666</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>B200, B200C, B200GT, B200CDT</td>
<td>Installation of Universal Avionics UNS-1E NGU, Config Module and SIS27JU.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/authorized noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Avionics Straubing modification rAsR-2013-051 Am 00 dated 15/05/2013.Operate in accordance with Avionics Straubing Flight Manual Supplement ASR-2013-051-FMS-05-00 dated 15/05/2013. Maintain in accordance with Avionics Straubing ICA rAsR-2013-051- ICA-11-00 Am 00...</td>
<td>No</td>
<td>10/07/2013</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<th>System Record</th>
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<tr>
<td>10045671</td>
<td>AMERICAN AIRLINES, Inc.</td>
<td>10045671</td>
<td>FAA STC ST014836C-O</td>
<td>757-200</td>
<td>Interior reconfiguration, new lavatories and In Flight Entertainment Equipment upgrade</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. For applicable regulations refer to FAA STC ST014836C-O. The requirements for environmental protection and the associated identified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List: American Airlines document No. F1470-MDL, Revision G, dated 9 October 2012-Instructions for Continued Airworthiness, American Airlines document No. F1470-ICA, Revision C, 28 March 2012-Aircraft Flight Manual Supplement, American Airlines document No. F1470-AMFS, Revision B, dated 9 October 2006.</td>
<td>None</td>
<td>11/07/2013</td>
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<td>10045671</td>
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<tr>
<td>10045675</td>
<td>ROSEN SUNVISOR SYSTEMS L.L.C.</td>
<td>10045675</td>
<td>FAA STC SA1950NM</td>
<td>CESSNA 500, 550, 555, 560</td>
<td>Rosen Sunvisor System Monorail mounted cockpit sunvisor</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated identified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rosen Sunvisor Systems, Drawing List R605-06DL, revision D, dated 22 January 2009-Rosen Sunvisor Systems, Installation Instructions for Cessna Clatteron Monorail Sunvisor System, KIR506S-010-S, Doc. 2641-0126-001, rev. D, date 21 January 2009-Rosen Sunvisor Systems, Installation Instructions for Cessna.</td>
<td>None</td>
<td>11/07/2013</td>
<td>Active</td>
<td>10045675</td>
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<td>10045732</td>
<td>AVIONITEC Ltd.</td>
<td>10045732</td>
<td>PA-34-200</td>
<td>Dual GNS upgrade to WAAS Units incl. LPV Certification/Compliance showing to AMC20-27 as the installed equipment is suitable for RNAV (GNS) Approaches down to LPV MINIMA Compliance showing to AMC20-27, resp. CR-3.01.10.A, (Airworthiness Criteria for RNAV Equipment supporting APV BARO VNAV approaches to...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificate noise etc. or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List: 1418-17, dated 28 April 2013-Instructions for Continued Airworthiness, 1418-015, dated 15 May 2013-Airplane Flight Manual Supplement: 1418-011, dated 23 February 2013-revisions of the above listed documents approved by EASA.</td>
<td>S/N: 34-7530164 HB-LQJ The approval holder shall fulfill the obligations of Part 21 §21A119 Prior to installation of this modification it must be determined if the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effects...</td>
<td>15/07/2013</td>
<td>Active</td>
<td>10045732</td>
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<tr>
<td>10045763</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>10045763</td>
<td>FAA STC STG044NY</td>
<td>777-200</td>
<td>Installation of Panasonic Avionics Corporation System 3000passengerentertainment system.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Certification basis for this type design change is Title 14 CFR Part 29 including amendments 25-1 to 25-112(less: equivalent of JAR 25 Change 13 for the following: 25...</td>
<td>Master Drawing List, Delta Engineering document No. 0143-05393-2074, revision E, dated 15 April 2006-Instructions for Continued Airworthiness, Delta Engineering document No. 0743-05399-2074, Revision A, dated 3 January 2005-revisions of the above listed documents approved by EASA in accordance with FAA STC STG04399, appropriately approved in accordance with Technical Implementation Procedures of EU/USA Bilateral Agreement, is required to be installed as a prerequisite for completion of this STC Prior to installation of this design change it must be determined if the interrelationship between...</td>
<td>FAA STC ST01493E, appropriately approved in accordance with Technical Implementation Procedures of EU/USA Bilateral Agreement, is required to be installed as a prerequisite for completion of this STC. Prior to installation of this design change it must be determined if the interrelationship between...</td>
<td>16/07/2013</td>
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<td>10045780</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>10045780</td>
<td>REV. 2</td>
<td>551 (CERTIFICATION II) 501 (CERTIFICATION I)</td>
<td>Single/Dual Installation of GPS/COM/NAV System Garmin GTN650/750</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Certification basis for this type design change is Title 14 CFR Part 29 including amendments 25-1 to 25-112(less: equivalent of JAR 25 Change 13 for the following: 25...</td>
<td>Flight Manual Supplement ASA-2013-018-FMS-05-00, Rev. 08 or later revisions of the above listed documents approved by EASA Instructions for Installation ASA-2013-018-FMS-10-00, Rev. 08 or later approved revisions/instructions for Continued Airworthiness ASA-2013-018-ICA-11-02 (Rev 02) dated 12th April...</td>
<td>The aircraft must have been already equipped with (a) suitable E-HSI(s). Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effects upon the air...</td>
<td>30/08/2017</td>
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<td>10045766</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td>561 AND 551</td>
<td>Installation of dual EHSI Sandel SN4500 and single E-ADI Sandel SN4550</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraphs at a later amendment: CS-23, Amendment 3 (not required by 21.A.101) 307, 561, 575, 603, 605, 607, 609, 611, 613, 771, 773, 863, 867, 1182, 1301, 1...</td>
<td>Flight Manual Supplements - ASR-2013-021 - FMS-05.1-00, Rev. 00- ASR-2013-021 - FMS-05.2-00, Rev. 00, other later revisions of the above listed documents approved by EASA. For installation ASR-2013-021 - IFL-10-00, Rev. 00 or later EASA approved revisions/instructions for Continued Airworthiness ASR-2013...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse interaction between this design change and any other previously installed design changes and/or repair will introduce a degrades effect upon the airworthiness of the product.</td>
<td>22/07/2013</td>
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<td>10045814</td>
<td>BOLDER MANUFACTURING G. L. C.</td>
<td>FAA STC ST03233CH</td>
<td>787-300BCF</td>
<td>Installation of Galley F5</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Bolder Manufacturing Master Data List: MDD- BMM0040D01M2, Revision B.16/04/01O and later revisions of the above listed documents approved by EASA and FAA EASA ED Decision 2004/05/01 (or subsequent revisions off this decision) and/or the Technical Implementation Procedures of EU/USA.</td>
<td>None</td>
<td>19/07/2013</td>
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<td>10045816</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC ST016968E</td>
<td>747-400</td>
<td>Interior reconfiguration of KLM B747-400 Aircraft Installation (Information Class Ottomans, a new Closet, new Class Dividers, modified Pilot Crew Rest, new Stowage Bin and new Upper Deck Decompression Venting)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: 14 CFR Part 25 at amendment 25-54-25-35(6)(2), (g)(25.571 (a))EASA-CS25 Amendment level 12.25.36(a)(6), (c)(5)(a)(25.5825.853 (a)(3)(f)(9) 25.869(c)(1)(a)/(3)/25.1529The requirement...</td>
<td>BIFPS202 D01, Revision J, 19 July 2013 - Master Drawing ListBIFPS202-D10, Revision B. 11 July 2013 - Certification PlanMPS202 D15, Revision IR, 09 May 2013 - Instructions for Continued Airworthiness Flight Manual Supplements: - FMS-05.2-00, Revision 00 or later revisions of the above listed documents approved by ASR-2013-021 - IFL-10-00, Rev. 00 or later EASA approved revisions/instructions for Continued Airworthiness ASR-2013...</td>
<td>This STC is a prerequisite of the KLM STC with project number01/01/31967E-001. The following STC’s should be installed simultaneously: STC ASR-2013-021-021 - FMS-05.1-00, Rev. 00- ASR-2013-021 - FMS-05.2-00, Rev. 00, other later revisions of the above listed documents approved by EASA. In EASA ED Decision 2004/05/01 (or subsequent revisions off this decision) and/or the Technical Implementation Procedures of EU/USA.</td>
<td>19/07/2013</td>
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<td>1040586</td>
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<td>10045818</td>
<td>ATI ENGINEERING SERVICES, LLC</td>
<td>FAA SR0244NN-D</td>
<td>S-92A</td>
<td>Night Vision Goggle (NVG) compatible Lighting System Revision 1: Modification of cockpit, cabin and external lighting system to be compatible with the use of night vision goggle in all phases of flight.Revision 2: Inclusion of additional NVIS configurations</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraphs at a later amendment: CS29 Amdt. 2; 29.129.25(a), 29.129.25(c), 29.777(a), 29.773(a)(2), 29.777.29.853(a)(3)(f)(9), 29.1301, 29.1303(a)(9)(3)/(2) are not applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement, Document No. FMS-100511.EASA, Rev. 1;Installation/Manufacturing Drawing List, Document No. IDL-100511, Rev. 21;Instructions for Continued Airworthiness, Document No. ICA-100511, Rev.11.Instructions for Continued Airworthiness, EASA Supplement S-1, DocumentNo.: ICA-1...</td>
<td>Installation of NVG/NVG compatible lighting does not include or imply approval for flight operations with Night Vision Goggles (NVG). The operator must receive approval from the Civil Aviation Administration to operate with NVG. Rotorcraft modified by this design change employs Night Vision Imaging...</td>
<td>06/06/2019</td>
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<td>10045827</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>ATR 42-500 ATR 42-300 ATR 42-200</td>
<td>22/07/2013</td>
<td>Prior to installation of this design change it must be determined that there is no adverse interaction between this design change and any other previously installed design changes and/or repair will introduce a degrades effect upon the airworthiness of the product.</td>
<td>Master Document list, ref: 20680525-DOC Issue 1, dated 19/07/2013.FMS Configuration with advisory VNAV - Aircraft Flight ManualSupplement, ref: 20680525-AM A Issue 2;FMS Configuration without advisory VNAV - Aircraft Flight Manual Supplement, ref: 20680525-AM B Issue 2;Instruction for Continued Airworthiness...</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10045832</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>REV. 8</td>
<td>777-300ER</td>
<td>First Class Cabin Refresh of Cathay Pacific Aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List: STC-2235D (NORTHWEST AEROSPACE TECHNOLOGIES, Inc.) - Installation of Bruce Aerospace 10045837</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2017</td>
<td>Active</td>
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<td>10045833</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>DC-9-83 (MD-83)</td>
<td>Cargo Seat Bag Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s): CRI D-01 -(0/10025850). The following paragraph(s) at a later amendment: CS 25.853 Amdt 12The requirements for environmental protection and the as...</td>
<td>RUAG AEROSPACE TECHNOLOGIES INC. ULS-1w FMS and Thormen Air Data System AD32 Installation:</td>
<td>- Master Document List: 411A-MLD Rev. N/C dtd. 18.07.2013 - AFM Supplement 411A-AMFS Rev. N/C dtd. 14.06.2013</td>
<td>per AFM Supplement 411A-AMFS Rev. N/C dtd. 14.06.2013</td>
<td>22/07/2013</td>
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<tr>
<td>10045837</td>
<td>RUAG AG</td>
<td>REV. 1</td>
<td>A109E</td>
<td>RUAG UNS-1w A109 Universal UNS-1w FMS and Thormen Air Data System AD32 Installation:</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement for A109E with Universal Flight Management System (FMS) UNS-1w and Thormen AD32 ArData Display Reference 1894714-50, Revision 002, dated 15.07.2013</td>
<td>S/N 11106 Prior to installation of these changes/repair it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/01/2021</td>
<td>Active</td>
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<tr>
<td>10045841</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST01172NY</td>
<td>Installation of Bruce Aerospace Emergency Path Lighting System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>ODA-2356-01, Revision A, 3 May 2013 - Project PlanSheet 09432-2350, Revision -</td>
<td>This STC is a prerequisite of the KLM STC with project number0101018346-001.Prior to installation of this design change must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/07/2013</td>
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<tr>
<td>10045892</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL</td>
<td>Introduction of TCAS 2000 v7 Revision 1 approves both ACASII change 7 and change 7.1 aspects of aTCAS 2000 system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- STC Twenty One Ltd Flight Manual Supplement No. S21.FMS-0096 Issue 2 (757-200 series aircraft) - STC Twenty One Ltd Flight Manual Supplement No. S21.FMS-0095 Issue 1 (757-300 series aircraft) - STC Twenty One Ltd Flight Manual Supplement No. S21.FMS-0094 Issue 4 (757-300 series aircraft) - STC Twenty One Ltd Flight Manual Supplement - ICA No. S21.TEC-0338 Issue 3- STC Twenty One Ltd Master Document List</td>
<td>This Revision 1 to EASA STC 10045892 is applicable to aircraft with EASA STC 10043170 installed and approves changes equivalent to those approved under FAA STCs SA3964SW-D and SA3235SO. This design change does not activate or approve ACAS II Hybrid Surveillance. Operational Limitations of the TCAS are co...</td>
<td>04/11/2019</td>
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<td>10045883</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SINC</td>
<td>REV 2</td>
<td>A300F4-605ST</td>
<td>Modification ST25MOD069 issue 1; &quot;Installation of a 6th seat in the cockpit&quot;. This approval covers also approval of the following modifications: (i) ST25MOD006 issue 3; &quot;Oxygen supply for 5th and 6th occupant&quot;; ST25MOD014 issue 1; &quot;FR10.7, reinforcement of the pressure bulkhead&quot;; ST23MOD015 issue 1; &quot;reinforcement of the pressure bulkhead. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (i) Special Condition (s): CRI D-301 Occupancy; (ii) The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as...</td>
<td>Airbus Transp. International AFM Supplement &quot;Appendix AF - Sixth Seat in the cockpit&quot; ST25MOD069&quot; dated 10 January 2014 for revisions of the above listed documents approved by EASA.</td>
<td></td>
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<td>20/05/2014</td>
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<tr>
<td>10045888</td>
<td>KLM ENGINEERING &amp; MAINTENANCE</td>
<td>747-400</td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (i) Special Condition (s): CRI D-GEN02 PTC - Application of heat release and smoke density requirements to seat materials - the following paragraph(s) at a later amendment: CS 2...</td>
<td>EA-018213, Revision 2, 10 July 2013 - Certification Plan/CA-747-01-213-001, Revision Org, 10 July 2013 - Instructions for Continued Airworthiness/MDL-012613, Revision 2, 25 July 2013 - Master Data List/EC-18552, Revision 2, 25 July 2013 - Compliance Check List/EA012613, Revision 5, 25 July 2013 - Decal...</td>
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<td>10045880</td>
<td>JET AVIONICS AVIATION &amp; MARINESERVICE CENTER S.r.l.</td>
<td>PA-31T CHEYENNE II</td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (i) Special Condition (s): CRI D-GEN02 PTC - Application of heat release and smoke density requirements to seat materials - the following paragraph(s) at a later amendment: CS 2...</td>
<td>Master Drawing List BDI 002-15-01 Ed. 1, Rev. 2 dated 02-06-13/MA MML-0017/01 Rev. 1. 17.04.2013/MS for GNS-5060, Doc. BDI 002-10-16 Rev. 0/5 MS for ST-3400, Doc. BDI 002-10-16 Rev. 0/5 FMS SN 3308, Doc. BDI 002-10-16 Rev. 0/5 MWEU Update BDI 002-10-17 Rev. 1</td>
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<td>25/07/2013</td>
<td>Active</td>
<td>10045880</td>
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<tr>
<td>10045881</td>
<td>AERODATA AG</td>
<td>B300</td>
<td></td>
<td>The Certification Basis for the original product remains applicable forirrelevant not affected by the modification, for items affected by themodification the following new Certification Basis is elected to comply with: CS-23 Amdt. 3, except 23.561 and 23.785 looked over. FAR 23.561 Amdt. 3 and FAR 23.785 Amdt. 3...</td>
<td>Installation in accordance with Master Document List, Aerodata Doc. No.MDL-A-C-840, Rev. 1, dated 19 July 2013/Airport Flight Manual Supplement, Aerodata Doc. No. AFM-134, Rev. 0, dated 05 June 2013/Aircraft Maintenance Manual Supplement, Aerodata Doc. No. AMM-1137, Rev.2, dated 15 July 2013 or later...</td>
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<td>10045899</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA P.</td>
<td>BD-100-1A10</td>
<td>Installation of a Digital Voice Check-List Management System in accordance with Master Document List BASS68010006 rev A dated November 3,2005</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unannounced and remain applicable to this certificate/approval.</td>
<td>- Bombardier Business Aviation Services BBAS Master Document List/BASS68010006 rev A. dated November 3, 2005 - Amtel Aeroaerical (AA) Flight Manual Supplement FM388400 Rev R dated 28 October 2005 - Instructions for Continued Airworthiness and supplemental maintenance information BAS68010006-9F</td>
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<td>26/07/2013</td>
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<td>EASA STC no.</td>
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<td>Date of Issue</td>
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<td>10045901</td>
<td>SAABEN TECHNICS BOD</td>
<td>REV 4</td>
<td></td>
<td>A318-112</td>
<td>Installation of cabin systems upgrade: Satellite communication system antennas and antennas-Satellite Live TV system-in-Flight Entertainment and Cabin Management system—New humidifier system—Removal of structural limitations</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-5310-01 “Glass Displays / Monitors”/Devarison(s): CRI D-5305-019 “Fire Handsfly”</td>
<td>Modification Approval Sheet MAS-S-00-2294, Issue 6, dated 29/02/2014</td>
<td>The approval is limited to MSN 29/10. Prior to installation of this modification it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/02/2014</td>
<td>Active</td>
<td>10045901</td>
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<tr>
<td>10045906</td>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td></td>
<td>737-900, 737-900ER, 737-800, 737-700, 737-800.</td>
<td>Activation of the GSM capability using the Triagnoys GSM Connexsulation.</td>
<td>The applicable certification basis (CB) is the one of the B737-900ER / B737-800 (25 change 15). The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List 01397-EAS-A-MDL rev A, Aircraft Flight Manual Supplement 01397-EAS-A-APM-01 revA/Notes for Continuous Airworthiness Supplement 01397-EAS-A-ICN-01rev Cor later revisions of the above listed documents approved by EASA.</td>
<td>- This approval is limited to the aircraft configurations identified in document 01397-EAS-A-AEL-01 (last revision). Airworthiness limitations as per 01397-EAS-A-ICA-01 rev C Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/07/2013</td>
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<td>10045910</td>
<td>EAD INC.</td>
<td>FAA STC ST0134CH</td>
<td>MD-11F</td>
<td>ELT Installation Installation of an ELTA ADT 406 AFAP or ELTA ADT 406 AFAP in Model MD-11F Emergency Locator/Transmitter.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>For Emergency Locator Transmitter ELTA ADT 406 AFAP in Model MD-11F - Certification Plan DC303-0670-00, Revision A, 26th August 2004.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/07/2013</td>
<td>Active</td>
<td>10045910</td>
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<tr>
<td>10045940</td>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>TCCA SA96-110</td>
<td>DHC-6 SERIES 300</td>
<td>Installation of PT6A-34 engines derated to 620 SHP on DHC-6 Series 300 in accordance with TCCA STC SA96-110.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change in requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>Flight Manual Supplement FM99B94/01, Issue 1, TCCA approved August 1, 1999/Maintenance Manual Supplement MM99B94/01, Issue 1, TCCA approved August 1, 1999/64 later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canada Bilateral Agre...</td>
<td>Limited to serial numbers 130, 210, 231 and subsequent. The holder must determine whether this design change is compatible with previously approved modifications if the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person write...</td>
<td>30/07/2013</td>
<td>Active</td>
<td>10045940</td>
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<tr>
<td>10045947</td>
<td>ST ENGINEERING AEROSPACE LTD.</td>
<td>A330-343</td>
<td>Air Asia X Airbus A330-343 Passenger Compartment Reconfiguration Certification Plan 395/ QAIX001, Air Asia X Airbus A330 Passenger Compartment Reconfiguration.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Certification plan: P395/ QAIX001, rev A, Master Drawing List: ref 39500000, rev 0 A / D-Asia X A330 Passenger Compartment Reconfiguration on Compliance Summary Report: ref P395/ QAIX001, rev 0 / C-ICA: P395/4ED01-01, rev 0 / later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. EEPMS approved through FAA STC ST02248AT and EASA ST...</td>
<td>31/07/2013</td>
<td>Active</td>
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List of EASA Supplemental Type Certificates

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<tr>
<td>10045945</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>SA365 N1</td>
<td>DHC-8-401/402</td>
<td>RDR-1400C Weather Radar installation.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CS29 Amdt 3 all paragraphs except – 29.337; FAR29 Amdt. 16, 1956: FAR29 Amdt 18. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>2852158-AM issue 3 (TMF SUPPLEMENT) (29.07.2013) or later revisions of the above listed documents approved by EASA. 2852166-DCC issue 1 (Certification documents list) (30.07.2013) 2852166-E0 issue 1 (Engineering Order) (30.07.2013)</td>
<td>2852166-JS issue 1 (–)</td>
<td>02/08/2013</td>
<td>Active</td>
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<td>10045954</td>
<td>N2 DEVELOPMENT Inc.</td>
<td>FAA ST02466A</td>
<td>DHC-8-301/-311/-314/-315</td>
<td>Installation of cabin interior noise reduction system in configuration “3” aircraft (see MDL referenced below).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List (MDL) N2002 revision 1 dated May 15th, 2013. Maintenance Manual Supplement N20023001 rev 002 dated March 27th, 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA Decision 2004/04CF (or subsequent revisions of this decision) and/or the Tec..</td>
<td>Prior to installation of the design change it must be determined that the applicable airworthiness change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/07/2013</td>
<td>Active</td>
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<tr>
<td>10045968</td>
<td>AVIONICS DESIGN SERVICES LTD.</td>
<td>TCCA STC SA 12-56</td>
<td>DHC-8-102/-103/-106/-201</td>
<td>Application of ORATEX aircraft covering system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List (MDL) 11077-1, Revision N/C. Maintenance must be in accordance with TCCA approved Avionics DesignServices Ltd. Flight Manual Supplement FMS 11077-1, Revision N/C. C. Maintenance must be in accordance TCCA app..</td>
<td>Suitability of provisions has been demonstrated for Apple iPad 1, 2 and 3 STCs.</td>
<td>01/08/2013</td>
<td>Active</td>
<td>1045968</td>
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<tr>
<td>10045970</td>
<td>LAMITZ-PRENA FOLEIN FACTORY GmbH</td>
<td>REV. 11</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Application of ORATEX aircraft covering system.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS 23.1, CS 23.29, CS 23.301, CS 23.303, CS 23.305, CS 23.307, CS 23.375, CS 23.603, CS 23.605, CS 23.301, CS 23.303, CS 23.305, CS 23.307, CS 23.375, CS 23.603, CS 23.605, CS 23.301, CS 23.303, CS 23.305, CS 23.307, CS 23.375, CS 23.603, CS 23.605, CS 23.301, CS 23.303, CS 23.305, CS 23.307, CS 23.375, CS 23.603, CS 23.605, CS 23.301, CS 23.303, CS 23.305, CS 23.307, CS 23.375, CS 23.603, CS 23.605</td>
<td>Application of ORATEX covering &amp; coating system. No AD/C-51-00-011 issue M, dated 5th March 2020. ORATEX Application Manual &amp; Airplane Maintenance Manual Supplement Doc. No. AD/C-51-00-011, dated 15th March 2020.</td>
<td>Prior to installation of this design change it must be determined that the applicable airworthiness change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/05/2020</td>
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<td>10045974</td>
<td>sABENA TECHNICS BOD</td>
<td>REV. 11</td>
<td>A330-201, A330-202, A330-220</td>
<td>IAGOS MODIFICATION = Integration of routine Aircraft measurements into a Global Observation System with two possible configurations: – IAGOS system activated – IAGOS system deactivated. The Revision 1 extended the applicability of the modification, by adding new MSN. The Revision 2 extended the a..</td>
<td>The Type and OSD Certification Basis (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval e..</td>
<td>Master Document List MDL S-00-2374-01 issue D. Aircraft Flight Manual Supplement AP48-S2 F0348-00-01 issue 11. - Instruction for Continued Airworthiness (CAI): F0348-00-05 issue D. later revisions of the above listed documents approved by EASA.</td>
<td>The modification is applicable to the following aircraft Serial Numbers: MSN 421, 421, 989, 1259, 657, 836, 636, 997 and 1835. Prior to installation of this change/repair it must be determined that the applicable airworthiness change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/06/2023</td>
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<td>10045983</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>A319-115</td>
<td>Airbus Corporate Jet Centre Modification CJ1620 - &quot;MASTER MOD CJ - RELIANCE&quot; Revision due to update of AFM-S</td>
<td>The Certification Basis for the original product is amended hereby adding additional or alternative airworthiness requirements as listed in the General ORI A-4301-019 &quot;AJJC Project Certification Basis&quot; Special Condition(s): CR H-01 - X-Airline EWISThe requirements for environmental protection and the...</td>
<td>- Master Data Approval Sheet, ref.: CJ1620, issue 2, dated 24th October 2013. - Aircraft Flight Manual - AFM Supplement, ref.: FAA-5 MOD CJ1620, dated 1st October 2013. - Maintenance Planning Document - MPD Supplement(s) - July 2010 - MPS-COM CJ1620, dated June 2010 (for later revisions of the above listed...</td>
<td>25/10/2013</td>
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<tr>
<td>10045993</td>
<td>TESTORI AERO SUPPLY S.R.L.</td>
<td>777-200ER</td>
<td></td>
<td></td>
<td>Y/C CUSHION FOAMS &amp; FABRIC SEAT COVERS - Replacement of Y/C CUSHION FOAMS &amp; FABRIC SEAT COVERS with new unadvisedly manufactured by Testori Aero Supply (TAS) on B777 AC (equipped with Recaro Seats.)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate approval.</td>
<td>- Master Data Document LIST DOC. TAS-UN-STC-MDL-01 REV 2 DATED 02/08/2013, - Maintenance Drawing List DOC. TAS-UN-STC-MDL-01 REV 0 DATED 12/05/2013 OR later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair introduces no adverse effect upon the airworthiness of the product.</td>
<td>02/08/2013</td>
<td>Active</td>
<td>10045993</td>
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<td>10045998</td>
<td>BVE AEROSPACE, Inc. - FSI</td>
<td>FAA STC STG21965-E D</td>
<td>A330-300</td>
<td>A330-200</td>
<td>Installation of an interior Reconfiguration ZONE A - FIRST AND BUSINESS CLASS Installation of new T/C seats, outboard closets forward of Door 3, and DEAR, rearrangement of existing outboard storage bins and ends...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate approval.</td>
<td>- Certification Plan AFS5904-D10-EASA, rev. A, MASTER DRAWING LIST, ref AFS5904-D01, rev AC, ICA AFS5904-D15, rev D, AFM Supplement, TP9A-4A4AFM, rev 2, FAA Letter 1505-13-226, dated 25th July 2010 or later revisions of the above listed document approved by EASA.</td>
<td>1) Prior to installation of this modification it must be determined that the relationship between this design change and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. 2) Incorporation of Airbus Service Bulletin A330...</td>
<td>05/08/2013</td>
<td>Active</td>
<td>10045998</td>
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<tr>
<td>10046006</td>
<td>MID-CANADA MOD CENTER</td>
<td>TCCA STC SA11-6</td>
<td>MF50</td>
<td>TrustNorth Avionics Express WiFi system installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate approval.</td>
<td>- Avions Design Services Ltd Master Drawing List MOL-1094 Rev C dated March 30, 2011 - Mid Canada Mod Center Flight Manual Supplement FMS1094-1 Rev B (+) dated March 30, 2011 - Mid Canada Mod Center Instructions for Continued Airworthiness/CA10094-1 Rev B, dated March 30, 2011 or later revisions...</td>
<td>None</td>
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<td>10046006</td>
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<td>10046011</td>
<td>ASPEN AVIONICS, Inc.</td>
<td>REV. 5</td>
<td>FAA SA10825SC</td>
<td>SEE FAA APPROVED MODELS LIST SEE FAA AML SA10825SC</td>
<td>Option 1 - Installation of EFD1000 Primary Flight Display Option 2 - Installation of EFD1000/EFD500 Evolution Flight Display/OptionSystem Option 3 - Installation of EFD1000 ES system</td>
<td>The Type and O&amp;S Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Equivalent Safety Findings: EISF-EO1-01-01-01 (FAA, ELOSSA209694A-S-1) EASA Certification Basis (Continuation) - The following paragraph(s) at a later a...</td>
<td>Option 1 - Installation of EFD1000 Primary Flight Display System Replacement with Aspen Avionics Installation document MOL-126-00, Revision C, dated 3/28/2008; Option 2 - Installation of EFD500 Primary Flight Display System Replacement document MOL-126-00, Revision B, dated 3/17/2008 or document A-01-175-00, Revision B, dated 3/17/2008...</td>
<td>Aircraft registered in countries under EASA authority are not eligible for installation of the CS125 &quot;Connected Gateway&quot; unit under FAA STC SA10825SC at this time. Activation of the Angle of Attack (AOA) indication is only permitted for aircraft with flaps. Installation of EFD1000 ES system is approved to...</td>
<td>21/12/2022</td>
<td>Active</td>
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**List of EASA Supplemental Type Certificates**

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<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST01547WI-D</td>
<td>7903800P1850071000XP</td>
<td>Modification of Rockwell Collins FMS-6000 System installation to provide B/SAS/LPV Capability</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval. Installation in accordance with Rockwell Collins Master Drawing List (MDL), W/AIS (SBAS) LPV in line 21 equipped with Hawker Aircraft rev RCA-0186-001 Revision A, dated August 10, 2009. Operation in accordance with Rockwell Collins Airplane Flight Manuals. Modifications: RCA-1024 (Hawker 750) Revision C, dated...</td>
<td>DBU-4000 and DBU-4100 Data Base Units are incompatible with the FMS-6000 (Flight Management Computer) installed by this STC. Loading of the FM database requires PCD-3000, CPAS, or DBU-5000. If DBU data loading capability is required or desired, Rockwell Collins Inc, STC ST01431WI-D (EASA.IM.A.01070)...</td>
<td>08/02/2013</td>
<td>Active</td>
<td>10046017</td>
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<tr>
<td>AIRCRAFT CABIN MAINTENANCE GmbH</td>
<td></td>
<td></td>
<td>Installation of refurbished Cabin Attendant Restraint System PR-NEPA-2121S-4-03-XX on 165 seats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval. Refer to Master Document List E/O No. 127-1003944 Rev. 0 (01/07/2013) or later revisions of the above listed documents approved by EASA.</td>
<td>As per Service Bulletin No. 127-1003944 SEB 06 00, latest revision. Prior to installation of this design change it must be determined that the interference relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness.</td>
<td>07/08/2013</td>
<td>Active</td>
<td>10046024</td>
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<td>AEROLITE AG</td>
<td>REV. 2</td>
<td>L45</td>
<td>EAS Master Document List</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval. Flight Manual Supplement, ref. FMS-250-0023000-501, Rev B, of Dec 19, 2013 (Noise Limitation Section (Chapter 4) included in Operations Manual OPA-250-0023000-501, Rev A, of Dec 19, 2013) and all subsequent lists of the above listed documents approved by EASA. Configuration is controlled by Master Document List...</td>
<td>Design approval is valid from and including SN 45-233 to and including SN 45-2000. Prior to installation of this design change it must be determined that the interference relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon...</td>
<td>21/02/2019</td>
<td>Active</td>
<td>10046038</td>
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<td>MACH TECHNICAL DESIGN S.L.</td>
<td>A119, AN119MKII</td>
<td></td>
<td>Installation of an External Multipurpose Liquid Injection System within the following main components: an external tank with attached pump fixed to the belly of the helicopter, a control box installed in the cockpit and a remote control anchored to the cyclic stick.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS/27 (Aircraft), 27.1, 27.25, 27.45, 27.65, 27.141, 27.143, 27.161, 27.171, 27.173, 27.175, 27.177, 27.251, 27.301, 27.303, 27.305(a), 27.307(a), 27.308, 27.321, 27.337, ...</td>
<td>Rotorcraft Flight Manual Supplement, ref. FMS-FAA-0312 Rev Original, dated 15/03/2013 or later revisions of the above listed documents approved by EASA. Configuration is controlled by Master Document List...</td>
<td>When the External Multipurpose Liquid Injection System is installed, there will exceed allowable Vne is limited to the value which results from subtracting 10 KIAS from the never-exceed speed established in EASA Specification for each combination of altitude and temperature. Prior to installation...</td>
<td>08/08/2013</td>
<td>Active</td>
<td>10046041</td>
<td></td>
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<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC SA01636WI-D</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Modification of a Rockwell Collins, Inc, VHF-4000 and VHF-4000 Installations</td>
<td>As defined on FAA STC SA01636WI-D, the requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval. Installation in accordance with Rockwell Collins, Inc. Master Drawing List RCA-0359-001 as listed in the ABL to this STC or later revisions of the above listed documents approved by EASA. In accordance with EASA EC 2004/04/CF (or subsequent revisions of this decision) and/or the Technical imp.</td>
<td>Prior to installation of this design change it must be determined that the interference relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third parties is not permitted...</td>
<td>08/08/2013</td>
<td>Active</td>
<td>10046045</td>
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<td>ASI INNOVATION SAS</td>
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<td>ASI-FAM-0010-01 - Replacement of the existing TBNL turret by the MX11 turret.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval. Supplement Manuel de Maintenance ASI-MPS-0010-03 du 25 juillet 2013, Supplement Manuel de Vid ASI-4FM-0010-0 du 24 novembre 2010 or later revisions of the above listed documents approved by EASA.</td>
<td>Limitation to F406 serial number 0090. Prior to installation of this design change it must be determined that the interference relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/08/2013</td>
<td>Active</td>
<td>10046050</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<tr>
<td>SABENA TECHNICS BOD</td>
<td>A319-133</td>
<td>Installation of a Cabin System Upgrade, In-Flight Entertainment system, Cabin refurbishing, Cabin communication system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification Approval Sheet MA5-S-00-2310, Issue 5, dated 07/09/2013. Master Drawing List MDL-S-00-2310, Issue 8, dated 06/08/2013. Instructions for Continued Airworthiness ICA-D9046-00-001, Issue A, dated 02/07/2013. Airplane Flight Manual Supplement AFM S-D9046-00-01, Revision 1, dated 02/06/2013.</td>
<td>The approval is valid for A319-133 CFM56-7B67 only. Prior installation of this design change must be determined. The change introduces no adverse effect upon the airworthiness of the product.</td>
<td>12/06/2013</td>
<td>Active</td>
<td>10046077</td>
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<td>NORTHERN AVIONICS S.R.L.</td>
<td>AW109SP</td>
<td>NAV 467 Installation of Cabin WLAN and Web connection in the helicopter AW109SP.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>JARFMS Supplement RPM NAV-4875 and later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the change introduces no adverse effect upon the airworthiness of the product.</td>
<td>13/08/2013</td>
<td>Active</td>
<td>10046079</td>
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<td>A&amp;M AVIONICS, SL</td>
<td>B737-600/700/800/900/990/ER</td>
<td>Boeing-CMC Class 3 EFB System Operating System installation B737NG.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Investigation Report AVI-00450-CIR Rev.0.0, dated 11/07/2013.</td>
<td>The installation of an EFB according to certificate 10044973 or 10045157 is a prerequisite for the installation of the above mentioned Operating System. Prior to installation of this design change it must be determined that the change introduces no adverse effect upon the airworthiness of the product.</td>
<td>14/08/2013</td>
<td>Active</td>
<td>10046097</td>
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<tr>
<td>A&amp;M AVIONICS, SL</td>
<td>T310R</td>
<td>Installation of Dual System Garmin GNS530W in accordance with Installation Bulletin B1 05/13</td>
<td>Other: 14 CFR Part 23 according to Certification Programme PC &amp; MC-05/13. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Bulletin AVI-00450-ICA Rev.0.0, dated 06/08/2013.</td>
<td>Approval limited to Airplane CESSNA T310R, S/N 310R-1408-Prior to installation of this design change it must be determined that the change introduces no adverse effect upon the airworthiness of the product.</td>
<td>14/08/2013</td>
<td>Active</td>
<td>10046100</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>MF50</td>
<td>F50 220V/50 Hz installation/Description as per Associated Technical Documentation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List AE13119MDL Rev. 1, dated 02/06/2013. Certification Program AE13119CPDP Rev. 2, dated 11/07/2013. Safety Analysis AE13119A Rev. 0, dated 03/06/2013. Electrical Loads Analysis AE13119E Rev. 0, dated 03/06/2013. Structural Analysis AE13119SA Rev. 1, dated 04/07/2013. Ground Test Program.</td>
<td>Prior to installation of this design change it must be determined that the change introduces no adverse effect upon the airworthiness of the product.</td>
<td>14/08/2013</td>
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<td>10046101</td>
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<td>STC Holder</td>
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<td>Date of Issue</td>
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<td>System Record</td>
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<td>COMMUTER AIR TECHNOLOGY Inc.</td>
<td>FAA STC SA2735CE</td>
<td>BEECH 100 AND 1100</td>
<td>Installation of External Lower Fuselage Cargo Compartment.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Devco Drawing List Number100DL, FAA approved dated July 2, 1991.Operate in accordance with FAA approved Flight Manual Supplement, dated July 29, 1991. For Beech Models 100 and 1100 with Commuter Air/Technology, inc. 60 Cubic foot external lower fuselage Baggage/ Cargo compartment.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third p...</td>
<td>14/08/2013</td>
<td>Active</td>
<td>10246102</td>
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<td>ENGINE TECHNOLOGIES, Inc.</td>
<td>FAA STC SA623NM</td>
<td>2. G36 1. V55B, A36TC, B36TC 1. F33A, G33, J35, V35A 1. 35-C33A, 36, A36, E33A</td>
<td>Installation of modified Teledyne Continental engine equipped with Turbo-normalizing System, optional Oxygen System and optional HarriettPHC- C37F-1RF/FS068 propeller.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Tomado Alloy Turbo Drawing List NumberTAT- BD700-001, Rev NC, dated September 29, 1999, or later FAA approved revision.Operate in accordance with Flight Manual Supplements as defined in Limitations and Conditions section or later revisions of the above listed documents ap...</td>
<td>See FAA approved Annex</td>
<td>14/08/2013</td>
<td>Active</td>
<td>10246106</td>
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<tr>
<td>CAV AEROSPACE Inc.</td>
<td>FAA STC SA623NM</td>
<td>2. G36 A. A36 (SERIAL E-2112 &amp; AFTER)</td>
<td>Installation of TKS fluid ice Protection System for flight into known icing conditions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Aerospace Systems &amp; Technologies (AS&lt;(&gt;&amp;&lt;) &gt;) Drawing List No 163000L, Revision 3 datedFebruary 20, 2003. Operation in accordance with Supplement to Pilots Operating Handbook/or FAA Approved Flight Manual, Document No 10300-FMS, dated February 06, 2003, or Supplem...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/08/2013</td>
<td>Active</td>
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<tr>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD</td>
<td>TCCA STC SA12-43</td>
<td>BD-700-1A11 BD-700-1A10</td>
<td>Installation, Aircraft Composition Modifications.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition(s): CRJ D-02 Issue 2, Side Facing Divers Equivalent Safety Finding (s): CRJ D-04 Issue 2, Emergency Exit Signs- Deviation(s): CRJ D-01 Issue 2, Doors Be...</td>
<td>Transport Canada STC No. SA12-43 Issue 1, dated 17 October 2012.Innotech Aviation Configuration Definition Record CER- BD700-001 Revision E.Innotech Aviation Airplane Flight Manual Supplement AMES- IA6000-2500Revision A.Innotech Aviation Instructions for Continued Airworthiness documentDOXX00X1000- ICA (...</td>
<td>EASA approved configurations are identified in Configuration Definition Record CER: BD700-001. The Wireless Local Area Network (WLAN) is intended to provide internet connectivity and email services to the airplane cabin using portable electronic devices (PEDs). Any other intended function of this equipm...</td>
<td>14/08/2013</td>
<td>Active</td>
<td>10246109</td>
<td></td>
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<tr>
<td>ALTRAN LAB/c AIRCAD</td>
<td>A330-243</td>
<td>A330-200 MSN 627 Cabin Reconfiguration - new seats and new IFE systeminstallation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Document List MSL-12007_Rev0, dated 24/07/2013.Instructions for Continued Airworthiness ICA-12007_Rev0, dated 07/02/2013.Aircraft Maintenance Manual Supplement AMM-12007_Rev0, dated 04/03/2012.or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/08/2013</td>
<td>Active</td>
<td>10246112</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10046122</td>
<td>STC Holder</td>
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<td>PA-31, PA-31-300, PA-31-350 (CHIFFTAIR) PA-31-325</td>
<td>Fuselage Camera Hole Cutout – Installation of fuselage camera hole including 20 inch opening and resulting of elevator trim and rudder trimmings.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Infinion Certification Engineering Inc. Installation Instructions Piper Navajo Fuselage Cutout (EO-114-111), Issue 8, dated 01 February 2019 Operation in accordance with Infinion Certification Engineering Inc. Flight Manual Supplement (PA-31, PA-31-300, PA-31-325, PA-31-35)…</td>
<td>09/12/2019</td>
<td>Active</td>
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<tr>
<td>10046138</td>
<td>AVIATION SUPPORT GmbH</td>
<td>3</td>
<td>BO105 S (VARIANTS CBS-4/5) BO105 D (VARIANTS DB-4/5) BO105 C (VARIANTS CB, CB-4/5)</td>
<td>Title: Camera Mount for BO105</td>
<td>The Certification Basis for the original product as amendedly the following additional or alternative airworthiness requirements: The following paragraphs at a later amendment: CS27 Amtd. 2: CS27.1,3,27.1, 27.5, 27.27, 27.29, 27.31, 27.45, 27.49, 27.51,27.55, 27.57, 27.71, 27.27…</td>
<td>- Flight Manual Supplement, ASE-91209, Camera Mount BO105, Revision 1 for later revisions of the above listed documents approved by EASA. - Projectbeschreibung, ASE-91306, Camera Mount BO105, Series PB, ASE-91306, Revision 0-1 dated 26th December 2014. - Liste der gültigen Dokumente ASE-91209, Camera.</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/04/2015</td>
<td>Active</td>
<td>10046138</td>
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<td>10046144</td>
<td>JET AVIATION AG</td>
<td>1</td>
<td>B737-700, 737-800</td>
<td>Cabin Insulation Improvement as Defined in BB73725801-500, REV A, Project Description and MDL BB73725801-001 Master Data List Rev. 1: Inclusion of CS 25.856 (a) compliance.</td>
<td>The Type and OSIS Certification Bases (CB) for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/ approval w/…</td>
<td>- Engineering Order, Doc. No. BB73725801-500, 258000, Rev. IR, 26.08.2013 - Master Data List, Doc. No. MDL BB73725801-001 Rev. D, 16.12.2021 - Analysis Report, Doc. No. JBSL-M-037800AN, Issue A, 03.12.2021 - Flammability Test Report, Doc. No. JBSL-M-037800FLR, Issue A, 03.12.2021 - Declaration of C…</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2021</td>
<td>Active</td>
<td>10046144</td>
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<tr>
<td>10046149</td>
<td>SPECIALIST AVIATION SERVICE &amp; MILITARY</td>
<td>2</td>
<td>CAAS, UK RESPONSIBILITY</td>
<td>MD900 (202 CONFIGURATION)</td>
<td>Rev. 0 – MD902 NVIS (PAS/MOD/88) – Modification of aircraft enable Night Vision Imaging system (NVIS) certification using Class B Night Vision Goggles Rev. 1 – Addition of white phosphorescent Night Vision Goggles In accordance with SAS/MOD/880 Rev. 2 – Administrative revision</td>
<td>The Certification Basis for the original product as amendedly the following additional or alternative airworthiness requirements: the following paragraphs at a later amendment: CS27 141, 171, 773, 777, 1301, 1309, 1321, 1381, 1383, 1461, 1501, 1525, 1529, 1581, 1583, 1585, plus FAA AC29.20 (Change 4)…</td>
<td>Revision 0-Flight Manual Supplements FMS/80-002 Issue A, FMS/80-002 Issue B, FMS/80-002 Issue B, FMS/80-007 Issue B, Maintenance Programme Supplement MS/80-008 Issue C, Drawing List D/GR-099 Iss…</td>
<td>Required Installation of EASA STC 10008937, Aerompany NVIS on MD900 Installation of NVIS/NVG compatible lighting does not include or imply approval for flight operations with Night Vision Goggles (NVG). The Operator must receive approval from the civil aviation national/authority to operate with NV…</td>
<td>02/03/2017</td>
<td>Active</td>
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<tr>
<td>10046155</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>3</td>
<td>A319-133 (CJ)</td>
<td>Interior Outfitting on A319-133/1a defined in the – Project Description, Doc. No. MDL BY01-2500, Rev. C, 12.08.2013 – Master Data List, Doc. No. MDL BY01-2500, Rev. C, 22.08.2013</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this Certificate/ approval Special Conditions: - CRI D-4302-015 Isolated Compartments: CRI D-4302-015 Improved Flammability Standard for thermal/accidental insulation…</td>
<td>- Master Data List, Doc. No. MDL BY01-2500, Rev. C, 22.08.2013 - AFM, Doc. No. AFMB-01-2500, Rev. E, 22.08.2013 - ICA Doc. No. ICA- BY01-2500, Rev. A, 21.06.2013 for later revisions of the above listed documents approved by EASA.</td>
<td>Galley G1 &amp; G2 (Includes mechanical installation and functional Moving) of Galley Door installed by the STC with the EASA Project Number Ref.: 02/01/2013/01-008: Door Status Indication, Frangibility and Decompression is part of this AMAC Cabin Interior Outfitting STC, for which this TV is issued and/or…</td>
<td>23/06/2013</td>
<td>Active</td>
<td>10046155</td>
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**Notes:**
- EASA has not certified the equipment for which these provisions are intended. You must get an additional EASA Approval to install such equipment. Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.
- The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.
- The Certification Basis for the original product as amendedly the following additional or alternative airworthiness requirements: the following paragraphs at a later amendment: CS27 141, 171, 773, 777, 1301, 1309, 1321, 1381, 1383, 1461, 1501, 1525, 1529, 1581, 1583, 1585, plus FAA AC29.20 (Change 4).
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<th>EASA STC no.</th>
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<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Information</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
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<tr>
<td>01043156</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>REV. 8</td>
<td>FAA ST02819CH</td>
<td>A320-233</td>
<td>Installation of labeled Interface Module (TIM) Electronic Flight Bag (EFB) system Rev. 5: among others, following changes were implemented: - Configuration J: additional optional interface connection for EFB's existing Quick Access Recorder (QAR) flight data recorders. - Contingency feature configuration. - Configuration J: additional optional interface connection for EFB's existing Quick Access Recorder (QAR) flight data recorders. - Contingency feature configuration.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchangeable. - Application of a new CCAR regulation.</td>
<td>Master Data List (MDL), Rosemount Aerospace document number 35104048B revision 3.</td>
<td>26/08/2013</td>
<td>Active</td>
<td>01043156</td>
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<td>01046157</td>
<td>ZODIAC AEROSAFETY SYSTEMS</td>
<td>REV. 1</td>
<td>TCCA SA90-281</td>
<td>DHC-8-301/311-314/315</td>
<td>Installation of Authentic Engine Inlet De-Icer Boot / Installation of Authentic Engine Inlet De-Icer Boot (PN: 51-SP-162-31-1)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Instructions for Continued Airworthiness Maintenance in accordance with EASA</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/10/2015</td>
<td>Active</td>
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<td>01046164</td>
<td>AVIATION FABRICATORS, Inc.</td>
<td>FAA STC SA261/F</td>
<td>cessna 208, 208b</td>
<td>Stretching assembly with restraints and portable Ferno unit can be installed on existing cabin seat track on either side of the aircraft between FS 158.00 and FS 238.00 in accordance with FAA STC SA261/F.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>This STC is limited to Cessna 208 and 208b cargo airplanes that have a right hand cabin entry door or the Cessna 208 and 208b cargo airplanes that have a left hand cabin entry door.</td>
<td>23/08/2013</td>
<td>Active</td>
<td>01046164</td>
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<td>01046177</td>
<td>FIELD AVIATION EAST, LTD.</td>
<td>REV. 2</td>
<td>TCCA SA13-36</td>
<td>DHC-8-301/311-314/315</td>
<td>Flight Crew, Zodiac MF20-517 Full Face Oxygen / Smoke Mask Upgrade (Introduction of KFMS)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/01/2014</td>
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<td>01046179</td>
<td>AIRBUS S.A.S.</td>
<td>A320-200</td>
<td>Airbus Corporate Jet Centre Modification CJ1467 - 'MASTER MOD CJ - RAPO (Cabin &amp; Systems Upgrade)'</td>
<td>Airbus Corporate Jet Centre Modification CJ1467 - 'MASTER MOD CJ - RAPO (Cabin &amp; Systems Upgrade)'</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements as listed in the General CRI A-320-200-009 “ACJC Project Certification Basis” Special Condition(s): CRI H.01 - CRI E.25 - Firm Handhold, CRI C.24 - Firm Handhold.</td>
<td>Modification Approval Sheet, ref.: CJ1467, Issue 1, dated 10 August 2015</td>
<td>This STC is limited to MSN 3725, MSN 4117 &amp; MSN 4796. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/08/2013</td>
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<td>Description</td>
<td>Certification basis</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>FOKKER SERVICES B.V.</td>
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<td>DHC-8-401, DHC-8-402 DHC-8-314, DHC-8-315 DHC-8-301, DHC-8-311 DHC-8-201, DHC-8-202 DHC-8-106 DHC-8-102, DHC-8-103</td>
<td>Provisions for a Class 2 Electronic Flight Bag (EFB)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Report CRR-0113, CRR-0138, CRR-0179. No later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/02/2013</td>
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<td>10046214</td>
<td>PMV ENGINEERING</td>
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<td>A330-341 / A330-321 A330-301 / A330-302 / A330-223 / A330-201 / A330-202</td>
<td>In-flight Entertainment System Installation on A330. This STC installs IFE System, developed by the IMS Company.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Investigation Report, ref.: SCA-00210-CRR Revision 5, dated 29 August 2013. Instruction for Continued Airworthiness, ref.: SCA-00101-ICA, dated 18 May 2013. No later revisions of the above listed documents approved by EASA.</td>
<td>1. ... - The IFE system installation, defined in this STC, is only valid for seat configuration identical to the one defined in PRM eng. Documented: SCA-00101-CRR Revision 5, dated 29 August 2013 ... This STC does not constitute the approval of the seat arrangement, installation, nor the implementation...</td>
<td>30/08/2013</td>
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<td>10046218</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>A321-212 / A320-214 / A320-111 / A321-211 / A320-115 / A321-115 / A320-112 / A321-112</td>
<td>Installation of HelioJet LED cabin light.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>... Supplemental Type Definition Document STD03-1319M Issue 1 Rev Dated 16/08/2013... Compliance Record List CRL3-1319M Issue 1 Rev 0 dated 16/08/2013... Instructions for Continued Airworthiness A-00-13-41-ICA Issue 1 Rev Dated 14/08/2013... Applicant’s Declaration of Compliance N. 3523 dated 16/08/2013...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/08/2013</td>
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<td>10046224</td>
<td>SAFRAN ELECTRONICS &amp; DEFENSEBUSINE SS UNIT - HELICOPTERS DATAMANAGEMENT NT SOLUTIONS &amp; SERVICES</td>
<td>REV. 1</td>
<td></td>
<td>SA 365 N, SA 365 N1 AS 365 N2, AS 365 N3</td>
<td>Installation of the Helcom V2 ANA embedded Equipment. The Helcom V2 ANA consists of a recording/transmitter equipment (flight data and system’s parameters). Rev. 1: STC holder name change from I.5.5/I International Services to SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT - HELICOPTERS DATA MANAGEMENT SOLUTIONS.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. - Special Condition: ON F-01 Lithium Battery. dated 20/03/2013. The requirements for environmental protection and the associated certified noise and/or emissions levels of...</td>
<td>Design Change Description N/3-AS365 ANA-HV2-ANA-E-06 dated 06/03/2013... Installation Manual NT-AS365 ANA-AS365-HV2-ANA-E-06 dated 06/08/2013... Maintenance Manual Supplement NT-AS365-ANA-AS365-ANA-E-A-06 dated 05/08/2013...</td>
<td>The change does not replace standard maintenance records and printout. - Any broadcasting must be done in accordance with National regulations... Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change...</td>
<td>27/02/2020</td>
<td>Active</td>
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<td>10046236</td>
<td>ALTRAN LABIA AIRCAD</td>
<td>REV. 1</td>
<td></td>
<td>MF50</td>
<td>Thera® &amp; Thera® TT-50064 ICA Satcom antenna installation Description as per Associated Technical Documentation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List Ref. MDL-13002 Revision 2, dated 31/07/2013... Certification Program CP-13002 Revision 1, dated 31/07/2013... ICA Manual ICA-13002 Revision 0, dated 10/07/2013... AMM Supplement AAM-13002 Revision 0, dated 03/07/2013... Change Approval Sheet CAS-13002 Revision 2, dated 23/08/2013 later revisions...</td>
<td>None</td>
<td>24/09/2013</td>
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<td>10046307</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>REV. 1</td>
<td></td>
<td>A340-313</td>
<td>Interior Reconfiguration</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. - Special Condition (s): D-07 Glass Screens of Displays / Monitors- Deactivation(s): D-08 Heat Release and Smoke Density The requirements for environmental protection and the...</td>
<td>-Master Data List MDL-DFP1-2500 Rev. B dated 6 September 2013 as amended by MDL-DFP2-2500 Rev. B dated 16 October 2014... Project Description Document PD-DFP1-2500 Rev. E dated 5 September 2013... AMF for supplement AFMS-DFP1-2500 Rev. D dated 9 October 2014...</td>
<td>-Limitations per AFM Supplement AFMS-DFP1-2500 Rev. G dated 9 October 2014... The installation of this STC is limited to MSN 526-Ths STC does not constitute the approval of installations defined by ICA Satcom STC ref. ST00965SC dated 10 March 2009 that have not been modified for which no new compliance...</td>
<td>22/10/2014</td>
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<td>EASA STC No.</td>
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<td>10046326</td>
<td>B/E AEROSPACE AVIATION MARINE S.A.R.L.</td>
<td>REV. 1</td>
<td>PA-44-180</td>
<td></td>
<td>Certification for P-RNAV operations and RNP LPV approaches of anestimating avionics configuration based on Garmin GNS 430W system and G500EFIS</td>
<td>SEAM Flight Manual supplement’s/’s N9MV/ 33688/11 version 2 dated 01/07/2013/SEAM modification file ‘s/’s 33811/11 revision 2 dated 01/07/2013 or later revisions of the above listed documents approved by EASA.</td>
<td>This STC is limited to aircraft w/ 44796308By this STC, the aircraft is approved for PRNAV, P-RNAV and RNP approaches operations with coupling of the Garmin GNS 430 W and G500 systems to the autopilot Autotrack IIC. Prior to installation of this design change it must be determined thatthe interrelationships between the design change and any other previously installed design change and/or repair will not introduce adverse effect upon the airworthiness of the product.</td>
<td>12/02/2013</td>
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<td>10046342</td>
<td>AIRLIFT A/S</td>
<td></td>
<td>AB139, AW139</td>
<td></td>
<td>Installation of SX-5 Search Light on Short Nose configuration (DO-139-33-102).</td>
<td>- Aircraft Flight Manual Supplement doc. FMS-139-33-102-01 Rev. 0 or later revisions of the above listed documents approved by EASA.</td>
<td>This STC is only applicable to AB1AW139 SN’s from 31005 up to 31200 (except 31007) and from 41001 to 41200. Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/or repair will not introduce adverse effect upon the airworthiness of the product.</td>
<td>10/09/2013</td>
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<td>10046397</td>
<td>FOKKER SERVICES B.V.</td>
<td></td>
<td>A321-231, A321-232, A321-211, A321-212, A321-213, A321-214, A321-215, A321-216, A321-111, A321-211, A321-212, A319-132, A319-133, A319-114, A319-115, A319-121, A319-112, A319-113.</td>
<td></td>
<td>Introduction of LEDs as a replacement of the existing Passenger/Compartments Tube Lights.</td>
<td>Certification Plan CP-0156 Issue 1/Compliance Record Report CRR-0156 Issue 1/Instructions for Continued Airworthiness Report SE-760 Issue 1 for later revisions of the above listed documents approved by EASA.</td>
<td>This STC is applicable to MSN 3151 and below only. Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/or repair will not introduce adverse effect upon the airworthiness of the product.</td>
<td>16/09/2013</td>
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<td>10046398</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
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<td>FAA STC ST04925E</td>
<td></td>
<td>Interior Reconfiguration installation of first class seats and ottomans, modified business class seats, VIP lavatories, seatbelt/cr -</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>The following STC’s must be installed concurrently with this STC: EASA STC 10045204 (FAA STC ST015148E) - Lavatory Lavatory G500 systems to the autopilot Autotrack IIC. Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/or repair will not introduce adverse effect upon the airworthiness of the product.</td>
<td>16/09/2013</td>
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<td>10046400</td>
<td>ANLUVI</td>
<td>REV. 2</td>
<td></td>
<td></td>
<td>Implementation of provisions for EFB Class 2 Provisions for obsolete Class 2 information system on both pilot stations Rev. 2. Transfer of STC</td>
<td>- Engineer Order and Certification Plan AM0005-46-20-EO. Rev. dated 05/04/2013-Master Data List AM0005-46-20-MDL. Rev. A, dated 11/07/2013/ Aircraft Flight Manual Supplement AM0005-46-20-AM. Rev. A, dated 05/03/2013-Instructions for Continued Airworthiness AM0005-46-20-CA. Rev. A, dated 05/03/2013-Cert...</td>
<td>This approval covers only the Class 2 EFB mounting devices and their power and data connection to the aircraft. This approval does not constitute an operational approval for the use of the EFB system. Prior to installation of this design change it must be determined that there is an interrelationship between...</td>
<td>05/05/2021</td>
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Notes:
- The list includes STCs for various models and modifications, including installations and reconfigurations.
- Certifications cover specific aspects such as environmental protection, noise, and operational capabilities.
- Each STC has associated technical documentation and limitations/conditions specified for its application.
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<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<td>10046408</td>
<td>CHALLENGER AVIATION PRODUCTS INC.</td>
<td>REV. 1</td>
<td>FAA STC SA01669CH</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Installation of Challenger re-usable air filter</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Challenger installation instructions 4402. Revision numbers per AML Instrutions for Continued Airworthiness 4003, revision D, dated August 31, 2005 or later revisions of the above listed documents approved by EASA in accordance with EASA and FAA STC Decision 2004/04 OF or subsequent revisions offer this decal...</td>
<td>24/02/2013</td>
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<td>10046455</td>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>REV. 2</td>
<td>A321-271N/-271NX/-272N/-272N</td>
<td>AS3/2543N/-2543N/-2543N/-2543N</td>
<td>Installation of Spectravox Dlink+ Controller Pilot Data Link Communication. Rev 2 includes A320 family NEO aircraft models and enables the utilisation of VHF 3 antenna position for datalink purposes.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition(s): SC F-0311-001 Flight Recorders including DataLink Recording. The following paragraph(s) at a later amendment: CS 25 Amendment 16 CS 25.1, CS 25.58...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/04/2021</td>
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<td>10046466</td>
<td>SPECIALIST AVIATION SERVICES LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A319-115</td>
<td>Internal Cargo Restraint System.</td>
<td></td>
<td>Flight Manual Supplement: FMS/629-003 Iss. A. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/09/2013</td>
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<td>10046488</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td>A318-115</td>
<td>Airbus Corporate Jet Centre – Modification CJ1200 – STC UPDATE FOR CABIN MODIFICATION &amp; DOOR LOCK INSTALL (Top Modification CJ1200 – VIP ProjectsBELAIR)</td>
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<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/03/2016</td>
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<td>Limitations/Conditions</td>
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<td>AEROSELE, Inc. - FSI</td>
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<td>STC02171SE</td>
<td>A340-600</td>
<td>Installation of Cabin Modification Provisions.Install new and modify existing electrical, structural and systems provisions only for the installation of a new cabin interior.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Plan BIFS115-D10, rev A - Master Drawing List, ref BIFS115-D01, rev N - ICA: Supplement To Instructions For Continued Airworthiness For That Installation Of Cabin Modification Provisions On Lufthansa A340-642/Airplanes, ref BIFS151-D5, rev WR - FAA Letter 1055-13-289, dated on 20th Sept...</td>
<td>1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. 2) ICA must be incorporated into the operator's ma...</td>
<td>25/06/2013</td>
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<td>AEROTEC &amp; CONCEPT</td>
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<td>STC 13099 - installation GTN650</td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements.The following paragraph(s) at a later amendment: Refer to AE130999AICRev. 1 for paragraph(s) at C23 and C3 The requirements for environmental protection and the associated...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AE1309999ICA Rev 0 dated 07/05/113 - Instructions for Continued Airworthiness AE130999DL Rev 2 dated 24/09/13 - Master Drawing List AE130999AV Rev 0 dated 26/09/13 - Aircraft Flight Manual Supplement</td>
<td>Limitation to PA-30 serial number 301251 and 301582.</td>
<td>27/09/2013</td>
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<td>10046517</td>
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<td>ROW 44, Inc.</td>
<td>REV. 1</td>
<td>ST02258A-O</td>
<td>757-200, 757-300</td>
<td>Installation of a Row 44 Wireless Broadcast System Components and System Activation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, Row 44 document No. SIE-346-701, Revision K, dated 27 May 2015 - Aircraft Flight Manual Supplement, Row 44 document No. SIE-346-604, Revision B, dated 15 January 2015 - Instructions for Continued Airworthiness, Row 44 document No. SIE-467-603(-&gt;),-&gt;) Revision E, dated 20 March 2015</td>
<td>This installation must be installed in conjunction with or after the Row44 Wireless Broadcast Antenna System Provision installation FAA STCST02904CH validated by EASA. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other...</td>
<td>15/09/2015</td>
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<td>SATRANS CABIN GERMANY GmbH</td>
<td>REV. 1</td>
<td>A340-600 DLH Galley Relocation Description - Summary: The change is related to the Galley Relocation in LH47 A340-600 aircraft for Q33, Q24, Q25, Q26, Q27 &amp; Pelmet Issue 1: ChangeRevision to confirm Service Bulletin with type design follower attachment of Galley Unit G23 and G24. The master document...</td>
<td>A340-600 DLH Galley Relocation Description - Summary: The change is related to the Galley Relocation in LH47 A340-600 aircraft for Q33, Q24, Q25, Q26, Q27 &amp; Pelmet Issue 1: ChangeRevision to confirm Service Bulletin with type design follower attachment of Galley Unit G23 and G24. The master document...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Summary: CS-A 2011-098A Rev. D - Type Definition Document 10DG-100921-25-07-08 Rev. D - ICA: ICAS- A-2011-098A Rev. Newer later revisions of the above listed documents/approved by EASA.</td>
<td>1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. 2) This modification must be implemented in conj...</td>
<td>14/02/2014</td>
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<td>B/E AEROSPACE, Inc. - FSI</td>
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<td>ST02170SE</td>
<td>A340-600</td>
<td>Lower Lobe Fixed Crew Rest Installation. The installation is comprised of a new lower deck crew rest, cargo loading equipment and installation kit.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements Special Condition (x): D-01: Lower Deck Crew Rest Compartment Installation The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>- Certification Plan BIFS102-D10, rev C - Master Drawing List, ref BIFS151-D5, rev P - ICA: Instructions For Continued Airworthiness, ref BIFS102-D15, rev B - AFM Supplement TP-11-1024AF, rev 2 - FAA Letter 1055-13-288, dated on 26th September 2013</td>
<td>1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. 2) ICA must be incorporated into the operator's ma...</td>
<td>27/09/2013</td>
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**EASA STC No.**

- B/STC02171SE
- STC 13099
- ST02258A-O
- STC 13099 - installation GTN650
- ST02170SE
- STC 13099 - installation GTN650
- ST02258A-O
- STC 13099 - installation GTN650
- ST02170SE
- STC 13099 - installation GTN650
- ST02258A-O
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>10046549</td>
<td>RUECKER AEROSPACE GmbH</td>
<td></td>
<td></td>
<td>DC-9-83 (MD-83)</td>
<td>MD 83 stowage of items in FWD cabin</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/ approval.</td>
<td>Ruwacker Aerospace Master Document List No. MDL-252-00043-000 Issue Aded 23 September 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>30/09/2013</td>
<td>Active</td>
<td>1044502</td>
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<tr>
<td>10046552</td>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 1</td>
<td></td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Cabin refurbishment Inc LOPA change.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/ approval.</td>
<td>328 Design Master Document List No. MDL-006201 Issue A dated 25 September 2013—or later revisions of the above listed documents approved by EASA.</td>
<td>27/09/2018</td>
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<td>10046554</td>
<td>ACS-NAI Ltd.</td>
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<td>TCCA C-LSA13-057D</td>
<td>CL-600-2B19</td>
<td>Cabin Interior Re-configuration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/ approval.</td>
<td>Transport Canada STC No. C-LSA13-057D Issue 1 dated 6th September 2013 ACS-NAI Ltd. Modification Data Summary H89000 Revision N/C or later approved revision ACS-NAI Ltd. Airplane Flight Manual Supplement H89001 Revision N/C or later approved revision. Instructions for Continued Airworthiness CRU7248.</td>
<td>Restricted to ASN 7248</td>
<td>30/09/2013</td>
<td>Active</td>
<td>1044504</td>
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<td>10046576</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td></td>
<td>A340-600</td>
<td>Cabin Interior Reconfiguration—Summary Cabin Configuration by replacing all Business- and Economy Class Seats and modification of several components, also the Inflight Entertainment System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/ approval.</td>
<td>- Certification Program CP00-122/2M Issue 1 rev 4 Supplemental Type Definition Document STD000-122/2M Issue 1 rev 4- ICA, ref A-00-1307-ICA, Issue 1 rev 0 Compliance Record List CRU00-122/2M Issue 2 at rev 0 or later revisions of the above listed documents approved by EASA.</td>
<td>1) Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product. 2) This modification cover the Cabin Certification...</td>
<td>01/10/2013</td>
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<td>10046582</td>
<td>BOGGI AERONAUTICS S.R.L.</td>
<td>REV. 1</td>
<td></td>
<td>AVANTI II AVANTI</td>
<td>Antenna installation in accordance with BOGGI S.R.L modification ref 18/POS05-10.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any im...</td>
<td>Installation in accordance with BOGGI S.R.L Master Document List (MDL) ref 18/POS05-10 MDL-02 Issue 2 Maintain in accordance with BOGGI S.R.L. Instructions for Continued Airworthiness (ICA) ref 18/POS05-10 SRL Issue 2 or later revisions of the above listed documents approved/ accepted under the EASA sy...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/07/2020</td>
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<td>EASA STC No</td>
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<td>Foreign/NAA STC ref.</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10046600</td>
<td>INTER-TEC AERO DESIGN AND CERTIFICATION LTD</td>
<td>REV. 2</td>
<td>A321-111/-112/01-232 A321-121/-122/111-112/112-211/- 112 A320-216/-211/-212/-232 121/-212/-214/- 215 A319-131/-120/-133 A319-121/-122/-111/-112/114/- 115</td>
<td>Installation of Pitch Aircraft Seating System Models QCC1300 and QCC12001xRev. 2. Transfer of STC holder from INTER-TEC SERVICES Ltd to INTER-TEC AERO DESIGN AND CERTIFICATION LIMITED.</td>
<td>The Type and OSL Certification Bases (CB) for the original product description applicable to this certificate/ approval. The requirements for environmental protection and the associated noise and emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification Plan O/PLTN 669 Issue 4, dated May 20, 2013. Certification Compliance Report TEN 669/1 Issue 1, dated 18 September 2013. Modification in Report and Declaration of Compliance TEN 669 Issue 1, dated 18 September 2013. Instructions for Continued Airworthiness TEN 669 Issue 1 dated 13 June 2013.</td>
<td>Prior to installation of this change/ repair it must be determined that the interaction between this change/ repair and any other previously installed change/ repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/12/2021</td>
<td>Active</td>
<td>10046600</td>
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<td>10046602</td>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td>REV. 1</td>
<td>PA-31 NAVAJO</td>
<td>Introduction of 120 amp. Alternators on Piper PA-31 - Removal of the existing Prestolite 70 Amp. Alternators and installation of an AirPartNo.20 Amp. Alternators model P324RN, the mounting on the engine remanufactured, since the new alternator is a direct replacement. The aircraft power wiring and prot...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>MODE00050_MDL Master Data List Rev. 2, dated 9 September 2013. MODE00050_AFMS Airplane Flight Manual Supplement MODE0050_CA Instructions for Continued Airworthiness dated 26 June 2013. Initial issue#MODE00050_WBS Weight &amp; Balance Information dated 23 July 2013, initial and later revisions of the above...</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/10/2013</td>
<td>Active</td>
<td>10046602</td>
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<td>10046612</td>
<td>ECT INDUSTRIES, SITE D'AUXERRE - BA</td>
<td>269C</td>
<td>Installation of a AGNAV GUAR differential GPS system for agricultural spraying operation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Bayo Flight Manual Supplement n°2001-04- AFMS-AG-NAV revision 3 dated 01/01/2013 or later revisions of the above listed documents approved by EASA Bayo STC Master List n°2011-04-1&amp; revision 4 dated 11/06/2013</td>
<td>The AGNAV GPS is not an aeromedical navigation system. It must be used only as an information system for agricultural spraying operations in day VFR conditions. Prior to installation of this design change it must be determined that the interaction between this design change and any other previously...</td>
<td>04/10/2013</td>
<td>Active</td>
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<td>10046655</td>
<td>TURK HAVA YOLARI TECNİK A.S TURKISH TECHNIC</td>
<td>A340-311</td>
<td>CABIN LAYOUT CONVERSION FROM DUAL CLASS 279 PAX TO SINGLE CLASS 331 PAX</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Design Declaration Form CB-250006-DD01-01, Dated 10/07/2013 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third p...</td>
<td>08/10/2013</td>
<td>Active</td>
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<td>10046680</td>
<td>AIR ALLIANCE GmbH</td>
<td>REV. 2</td>
<td>35A</td>
<td>Replacement of Mechanical HSI by Sandel SN4500 EHSRevisions of this STC corrects equipment Part Numbers installed. Revision 2 of this STC adds the MSN 35A-111 to the approved list of aircrafts.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Air Alliance Master Document List: 2013_004_L33S Form2, revision 04, dated 09 October 2015 Air Alliance, AFM Supplement: 2013_004_L33S_5006, revision 04, dated 12 October 2015 Air Alliance, Instructions for Continued Airworthiness: 2013_004_L33S_5007, revision 01, dated 07 October 2015 or later revisions...</td>
<td>This approval is limited for MSN 35A-346 and 35A-111 only. Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the...</td>
<td>20/06/2016</td>
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<td>EASA STC No.</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10046675</td>
<td>DOA21LIMITED</td>
<td>REV. 1</td>
<td>737-700</td>
<td></td>
<td>Introduction of Corporate VIP Interior Rev. 1: Transfer of STC holder role to DOA21 Ltd.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s)D-02 Installed ComponentsD-05 Glass Screens of Displays and MonitorsD-09 Injury Criteria for Side-Facing DivanD-08 Shower InstallationD-...</td>
<td>STC Twenty One Ltd Master Document List No. S21-MDL-1081 Issue 2STC Twenty One Ltd Flight Manual Supplement No. S21-FMS-0100 Issue 1 for later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/10/2023</td>
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<td>10046679</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>FAA STC ST03229CH-D</td>
<td>BD-700-1A10, BD-700-1A11</td>
<td>Cobham Aviator 700D SATCOM Upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA STC No. ST03229CH-D dated 18 June 2013.Jet Aviation St.Louis, Inc Master Data List 35-354G001 Rev A, dated 16June 2013.or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (correspondent revisions of this decision) and/or the Technical...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/10/2013</td>
<td>Active</td>
<td>10046679</td>
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<td>10046684</td>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>FAA ST04132AT</td>
<td>767-300</td>
<td>Floor Reinforcement in a Boeing 767-300 Series Aircraft</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: CS52 at Amendment 13-25-623, 25-625: The requirements for environmental protection and the associated certified noise and/ or emissions levels of...</td>
<td>TIMCO Aerosystems Master Data List: 12T304R009,Revision IR, 04/10/2013TIMCO Aerosystems Instructions for Continued Airworthiness: 12T304R009,Revision IR, 06/06/2013or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revision).</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/10/2013</td>
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<td>10046686</td>
<td>ALPINE AEROTECH Ltd.</td>
<td>TCCA SH97-80</td>
<td>212, 412, 412EP</td>
<td>Installation of Alpine Aerotech Ltd. Exhaust Ejector/Deflector Assembly</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Approval Report - Manufacture of Alpine Exhaust ejector &amp; deflectors/assemblies 1918-07-713s, Rev. N (it includes Cert. Basis, CCL, MasterDrawing List D191M-96-048, Rev. B).or later revisions of the above listed documents approved by EASA inaccordance with the Technical Implementation Procedures of EU.</td>
<td>This approval is limited to Bell helicopter models 212, 412and 412EP the model 412CF is not currently EASA certified. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/10/2013</td>
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<td>10046686</td>
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<td>10046687</td>
<td>DV AVIATION, Inc. DBA NORTHPOINT AVIATION</td>
<td>FAA STC SA02271SE</td>
<td>B) 206H, 206H A, TU206G, L096G 206H, L096G, 206H, TU206G</td>
<td>Fabrication and installation of Sloyi wing recumbent LED landing and taxiight idstressors. 3. Transfer from &quot;Fifty-Ten, Yankees, LLC&quot; to &quot;DV Aviation, Inc., Sloyi, LLC. Master Drawing List 900-100, Revision B. dated 14 January 2013.Sloyi Service instructions for Continued Airworthiness (ICA), DocumentNumber 100-01, dated 14 January 2013.Pilot’s Operating Handbook and Airplane Flight Manual Supplement (AFMS).Document Number S30023A, Revision 0, dated 28 Ma...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/06/2023</td>
<td>Active</td>
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The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis for the MBT BK117 C-2 is applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis for the aircraft is applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis for the aircraft is applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

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<th>STC holder</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of issue</th>
<th>Status</th>
<th>System Record</th>
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<td>DOA21 LTD</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: JAR 25 Change 14: §583(a)(4) &amp; Appendix F Part 14. The requirements for environmental protection and the associated certified noise and emissions...</td>
<td>Certification Summary: ref : N0425-D00 rev A, dated 21 October 2013 Master Drawing List, ref : N0425-D10 rev D, dated 15 October 2013 Instruction for Continued Airworthiness, ref : N0425-D70 rev C, dated 14 October 2013.</td>
<td>prior to installation of this design change it must be determined that the installation of this modification...</td>
<td>22/10/2013</td>
<td>Active</td>
<td>10040787</td>
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<td>A340-311 / -312 / -313 Interior reconfiguration of Lufthansa A340 aircraft.Description : Interior cabin upgrade of Airbus A340-300 Lufthansa Airlines (LH) aircraft. The main cabin items modified by the modification are the following: - Removal of existing passenger seats, - Removal of various passenger cabin monuments (gal...</td>
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<td>22/10/2013</td>
<td>Active</td>
<td>10040787</td>
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<td>CEESSNA DUESSELDOF FICATION CENTER GMBH</td>
<td>P-RNAV Approval of UNS-1L</td>
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<td>DOA21 LTD</td>
<td>The Type and OSO Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): CRI D-03 Isolated Compartments, CRI D-04 Glass Screens of Displays and Monitors, CRI D-05 Shroud Installation: Deviation(s): CRI D-02 Firm Hand...</td>
<td>- Master Data List, STC21 document ref : S21.MDL-1030 issue 2, Aircraft Flight Manual Supplement, STC21 document ref : S21.FMS-0103 Issue 1, Supplement Manual, STC21 document ref : S21.TEC-0399 issue 1, dated 17th October 2013 (including Instructions for Continued Airworthiness) later revisions of the above.</td>
<td>Prior to installation of this design change it must be determined that the installation of this modification...</td>
<td>23/10/2013</td>
<td>Active</td>
<td>10040793</td>
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<td>365 AEROSPACE DQS LIMITED</td>
<td>767-300 767-300 Concorde Cabin LED Lighting Upgrade. 2. Transfer of STC from 365 Aerospace Ltd. to 365 Aerospace DQS Ltd. 3. Certification basis correction.</td>
<td>AEROECOD Modification Summary No : MOD-1338 Revision A : (+)= inserted 24th October 2013</td>
<td>Prior to installation of this design change it must be determined that the installation of this modification...</td>
<td>20/11/2023</td>
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<td>10040805</td>
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<td>AMES - AEROSPACE AND MECHANICAL ENGINEERING SERVICES INC., WALTERSTARZA CHER LEAMAN</td>
<td>B767-300 Cabin reconfiguration incl. IFE, ISPS and Seat Integration.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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| AIRCRAFT CABIN MAINTENANCE GmbH | 10046809            | 737-600, 737-600ER, 737-700, 737-800, 737-800, 737-800, 737-800, 737-900, 737-900, 737-900, 737-200, 737-200C | Exchange of webbing of Passenger Swat Belt PIN 2000-1-511-8220 PIN EPA 2662134-2134-55 for installation on Boeing 737 | Refer to Master Document list E/O No. 129-1004565 Rev. 0 (09.09.2013) or later revisions of the above listed documents approved by EASA. | Refer to Master Document list E/O No. 129-1004565 Rev. 0 (09.09.2013) or later revisions of the above listed documents approved by EASA. | - 
| LUFTHANSA TECHNIK AG         | 10046814            | A340-300 2 Class K Retrofit. Description - new IFE, Galley, Skytrays and Carpet. | Exchange of webbing of Passenger Swat Belt PIN 2000-1-511-8220 PIN EPA 2662134-2134-55 for installation on Boeing 737 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Exchange of webbing of Passenger Swat Belt PIN 2000-1-511-8220 PIN EPA 2662134-2134-55 for installation on Boeing 737 | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Exchange of webbing of Passenger Swat Belt PIN 2000-1-511-8220 PIN EPA 2662134-2134-55 for installation on Boeing 737 | 25/10/2013 | Active | 10046723 |
| SOCIETE AIR FRANCE S.A.       | 10046817            | A330-243                                                               | A330-243-2 Class K Retrofit. Description - new IFE, Galley, Skytrays and Carpet. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - 
| AEROTECH & CONCEPT           | 10046820            | F900DX                                                                | F900DX In-flight Entertainment system installation                           | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - 
| OGMA, INDUSTRIA AERONAUTICA DEPORTUGAL, S.A. | 10046826            | EMB-135BJ                                                              | EMB-135BJ In-flight Entertainment system installation                       | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | - 
<p>| DELTA ENGINEERING CORPORATION | 10046843            | A321-213-231-232, A321-111-112-112-121-211-212, A320-2131-232-233, A320-111-112-212-213, A319-111-112-132, A319-111-112-112-112-114-115 | Ownership transfer of EASA STC 10046843 This modification supports the compatibility of aircraft equipment with Cellular and Wireless Data Technologies. This Revision 1 identifies the change of STC ownership. | The Certification Basis for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | The Certification Basis for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | The Certification Basis for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | 28/06/2018 | Active | 10046826 |</p>
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<th>EASA STC No.</th>
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<th>Rev. Level</th>
<th>Foreign/NAA STC No.</th>
<th>Model</th>
<th>Description</th>
<th>Certification Basis</th>
<th>Associated Technical Document</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>10046850</td>
<td>ANORA INTERNATIONAL LLC</td>
<td>REV. 1</td>
<td>FAA STC 10046867</td>
<td>ATR-72-201/-202/202 ATR72-201/-202/211/-212</td>
<td>Mechanical Cargo Loading System installation for aircraft in crosssection with large cargo door implemented.Rev 1: Applicability list extended to new models</td>
<td>The Certification Basis for the original product as amed with additional or alternative airworthiness requirements JAR 25 Change 15 for 25.80.25.102.25.102.25.303.25.305.25.307.25.308.25.601.25.603.25.605.25.607.25.690.25.691.25.692.25.754.25.80.25.953 &amp; 25.601.25.603.25.605.25.607.25.690.25.691.25.692.25.754.25.80.25.953</td>
<td>FAA STC 10046873</td>
<td>30/06/2020</td>
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<td>10046867</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 1</td>
<td>FAA STC 10046868</td>
<td>A330-243, A330-343</td>
<td>INTRODUCTION OF FLEET STD FWC Panasonic eXPhone System activation on NORTHWEST Limitations/Conditions ANCRA 737-800 Installation of Zodiac NuLook Interior on 767-300 Active Associated Technical</td>
<td>The Certification Basis for the original product as amed with additional or alternative airworthiness requirements CS 25 Amendment 13 CS 25.1009 CS 25.1301 CS 25.1330 CS 25.1335 CS 25.1529 CS 25.1585 CS 25.1707 CS 25.1709 CS 25.1711 CS 25.1713 CS 25.1719 CS &amp; 25.1721 CS</td>
<td>Master Data List AES-330-2874MDL Issue 3</td>
<td>Rev 1</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/04/2021</td>
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<td>10046888</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 3</td>
<td>FAA STC 10046869</td>
<td>A340-642 / 643</td>
<td>Panasonic eXPhone System activation on A340-600 This STC installs a Panasonic eXPhone System allowing, in coordination with other devices, the use of cellphones during flight.</td>
<td>The Certification Basis for the original product as amed with additional or alternative airworthiness requirements CS 25 Amendment 4: 25.23.25.301.25.302.25.303.25.305.25.307.25.308.25.601.25.603.25.605.25.607.25.690.25.691.25.692.25.754.25.80.25.953 &amp; 25.1301.25.1303.25.1305.25.1307.25.1309.25.1310.25.1311.25.1312.25.1314.25.1315.25.1317.25.1360.25.1431 &amp; 25.143</td>
<td>Master Data List, Ref. : 0634-01-A-2103-R01</td>
<td>FAA AD 2005-20-05 must have been complied with prior to installation of this STC. FAA AD 2005-20-05 must have been complied with prior to installation of this STC.</td>
<td>11/06/2016</td>
<td>Active</td>
<td>10046868</td>
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<td>10046873</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC</td>
<td>REV. 1</td>
<td>FAA STC 10046870</td>
<td>767-300</td>
<td>Installation of Zodiac NuLook Interior on Condor aircraft.</td>
<td>The Certification Basis for the original product as amed with additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: JAR 25.613(j)(y) at Amendment 11, JAR 25.853(a)(i) at Amendment 11, JAR 25.1529 at Amendment 11, AIRAC2015(3) at Amendment 11. The requireme...</td>
<td>Certification Summary, NorthWest Aerospace Technologies, Inc. documents: N0561-D90, Revision -, dated 14 February 2018., - Instructions for Continued Airworthiness, HealthTechnician document No.HTM142(&lt;,&gt;) at Amendment 11, JAR 25.613(j)(y) at Amendment 11, JAR 25.853(a)(i) at Amendment 11, JAR 25.1529 at Amendment 11, Aircraft Flight Manual Supplement, Rev: 0634-01-A-2351-R01, dated 19 October 2013</td>
<td>FAA AD 2005-20-05 must have been complied with prior to installation of this STC.</td>
<td>15/02/2018</td>
<td>Active</td>
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<td>10046910</td>
<td>JET AVIATION AG</td>
<td>REV. 3</td>
<td>FAA STC 10046911</td>
<td>737-800</td>
<td>Transfer STC 10046910 Rev. 2 to 737-800 Cabin Interior Outfitting as defined in Project Description, Doc. No. BBJ37683-500 Rev. 1, dated 21.03.2012 Master Data List, Doc. No. BBJ37683-001 Rev. B, dated 01.11.2013</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. Special Condition(s) CRI D-02 Isolated Compartments, Issue QCRI D-02 Glass Switches of Displays / Monitors, Issue QCRI D-14 Shower Installation, Issue QCRI D-01 Installation of Interf...</td>
<td>FAA Approved Ancra International LLC Controlled Data List (CDL) CDL42311(&lt;,&gt;) at Revision H, dated May 13, 2019, or later FAA approved revision Ancra International LLC. Weight and Balance Manual Supplement, No. 9559(4)&lt;(,)&gt; at Issue D dated October 17, 2013, or later FAA approved revision Ancra International...</td>
<td>13/12/2017</td>
<td>Active</td>
<td>10046910</td>
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</table>
**List of EASA Supplemental Type Certificates**

- **10046919**
  - STC Holder: ANVI, Inc.
  - Rev: REV. 2
  - Description: Installation of an ANVI Inc. Exceedance and Trend Monitoring System to PMN ETM1000. Purpose of the STC revision 01 is the application of Exceedance and Trend Monitoring System to Bell 204B and Bell 206A-1. Purpose of this STC revision 02 is the application of Exceedance and Trend Monitoring System.
  - Certification Basis: The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated capability noise/and or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.
  - Limitations/Conditions: The Exceedance and Trend Monitoring System can only be installed on any FVFR aircraft with non-FADEC equipped engine (unless a specific Modified FADEC rotorcraft is noted in the AM). It is for ground crew interoperability maintenance functions and not intended to replace the original indicator as certified.

- **10046964**
  - STC Holder: WACO CLASSIC AIRCRAFT CORPORATION
  - Rev: REV. 1
  - Description: Modify the WACO YMF to a Classic "WACO FS" or "WACO F5C" configuration according to Classic Drawing Number CAC-FS.
  - Certification Basis: The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.

- **10046966**
  - STC Holder: JET AVIATION AG
  - Rev: REV. 3
  - Description: Passenger Gasous Oxygen System Rev. 2. Transfer STC 10046965. Rev. 3: Modification of the oxygen system due to part obsolescence and service support from part OEM. The modification comprises of alternative pulse latching valves, removal of oxygen regulators and limitators.
  - Certification Basis: The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment of C5.869 Amdt. 06, CS 19.1301 Amdt. 02, CS 25.1300 Amdt. 06, CS 25.1411 Initial Issue, CS 25.1453 Amdt. 04.

- **10046967**
  - STC Holder: DELTA ENGINEERING CORPORATION
  - Rev: REV. 1
  - Description: Panasonic Global Communications Subinstallation of a Panasonic Global Communications Suite Ku-Band Antenna.
  - Certification Basis: The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated capability noise/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

- **10046968**
  - STC Holder: DELTA ENGINEERING CORPORATION
  - Rev: REV. 2
  - Description: Panasonic Global Communications Subinstallation of a Panasonic Global Communications Suite which provides passengers with in-flight Wi-Fi, cellular phone and TV.
  - Certification Basis: The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated capability noise/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

- **10046969**
  - STC Holder: JET AVIATION AG
  - Rev: REV. 3
  - Description: Passenger Gasous Oxygen System Rev. 2. Transfer STC 10046965. Rev. 3: Modification of the oxygen system due to part obsolescence and service support from part OEM. The modification comprises of alternative pulse latching valves, removal of oxygen regulators and limitators.
  - Certification Basis: The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment of C5.869 Amdt. 06, CS 19.1301 Amdt. 02, CS 25.1300 Amdt. 06, CS 25.1411 Initial Issue, CS 25.1453 Amdt. 04.

- **10046970**
  - STC Holder: JET AVIATION AG
  - Rev: REV. 3
  - Description: Passenger Gasous Oxygen System Rev. 2. Transfer STC 10046965. Rev. 3: Modification of the oxygen system due to part obsolescence and service support from part OEM. The modification comprises of alternative pulse latching valves, removal of oxygen regulators and limitators.
  - Certification Basis: The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment of C5.869 Amdt. 06, CS 19.1301 Amdt. 02, CS 25.1300 Amdt. 06, CS 25.1411 Initial Issue, CS 25.1453 Amdt. 04.

- **10046971**
  - STC Holder: JET AVIATION AG
  - Rev: REV. 3
  - Description: Passenger Gasous Oxygen System Rev. 2. Transfer STC 10046965. Rev. 3: Modification of the oxygen system due to part obsolescence and service support from part OEM. The modification comprises of alternative pulse latching valves, removal of oxygen regulators and limitators.
  - Certification Basis: The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment of C5.869 Amdt. 06, CS 19.1301 Amdt. 02, CS 25.1300 Amdt. 06, CS 25.1411 Initial Issue, CS 25.1453 Amdt. 04.

- **10046972**
  - STC Holder: JET AVIATION AG
  - Rev: REV. 3
  - Description: Passenger Gasous Oxygen System Rev. 2. Transfer STC 10046965. Rev. 3: Modification of the oxygen system due to part obsolescence and service support from part OEM. The modification comprises of alternative pulse latching valves, removal of oxygen regulators and limitators.
  - Certification Basis: The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment of C5.869 Amdt. 06, CS 19.1301 Amdt. 02, CS 25.1300 Amdt. 06, CS 25.1411 Initial Issue, CS 25.1453 Amdt. 04.

- **10046973**
  - STC Holder: JET AVIATION AG
  - Rev: REV. 3
  - Description: Passenger Gasous Oxygen System Rev. 2. Transfer STC 10046965. Rev. 3: Modification of the oxygen system due to part obsolescence and service support from part OEM. The modification comprises of alternative pulse latching valves, removal of oxygen regulators and limitators.
  - Certification Basis: The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment of C5.869 Amdt. 06, CS 19.1301 Amdt. 02, CS 25.1300 Amdt. 06, CS 25.1411 Initial Issue, CS 25.1453 Amdt. 04.
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<th>STC Holder</th>
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<th>Associated Technical Documentation</th>
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<td>STC Holder 10046997</td>
<td>SABENA TECHNICS BCC</td>
<td>A319-111</td>
<td></td>
<td>Cabin modification including installation of structural provisions for monumonments installed in an A319.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List 4547-01-AT-2301-R00, dated 10th of October 2013; Airborne Flight Manual Supplement 4547-01-A-2301-R00, dated 13th of November 2013; later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft product.</td>
<td>19/11/2013</td>
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<td>STC Holder 10046998</td>
<td>SABENA TECHNICS BDO</td>
<td>A319-131</td>
<td></td>
<td>Cabin modification - Quick change</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Plan PC-9444-00-001, Issue 3 dated 06/11/2013; Modification Approval Sheet 25-25-2308, Issue 6 dated 23/10/2013; Master Drawing List MDL-S-25-2308-03, Issue B dated 23/10/2013; Instructions for Continued Airworthiness ICA-0944-25-...</td>
<td>Prior to installation of this design change it must be determined that the design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft product.</td>
<td>07/11/2013</td>
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<td>STC Holder 10046996</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>P2006T</td>
<td></td>
<td>provisions for magnetic field measurement (Step 1) and improvement of flight operational stability (Step 2)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (the following paragraph(s) at a later amendment: CS 23.1, Amendment 2; CS 23.23, Amendment 2; CS 23.24 (as), Amendment 2; CS 23.45, Amendment 2; CS 2...)</td>
<td>Engineering Order Doc. No. STC-AT-056, Rev. 5, dated 21.10.2013; (-&gt;) AFM Supplements Doc No. AFM1-1-2306-056, Rev. 2, dated 01.10.2011; (required for step 1 of this STC) and AFM &amp; STC-AT-056, Rev. 0, dated 05.04.2013 (required for step 2 of this STC)</td>
<td>Prior to installation of Step 1 of this STC, the following modifications have to be installed: Airborne Technologies Modification: STC-AT-003, Step 1 or 3 or 5 or 7; latest effective revision - Airborne Technologies Modification: STC-AT-003, Step 9; latest effective revision - STC-AT-056 Step 2 (Prior to...</td>
<td>08/11/2013</td>
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<td>STC Holder 10046995</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST0911NY</td>
<td>Installation of a Panasonic In-Flight Entertainment SystemRevision 1 adds the 777-200LR model.</td>
<td>The Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: (the following paragraph(s) at a later amendment: CS 23.1, Amendment 2; CS 23.24 (as), Amendment 2; CS 23.45, Amendment 2; CS 23.46, Amendment 2; CS 2...)</td>
<td>1. Delta Engineering Master Data List 0143-0848-2348 Revision A2 dated 29 September 2021; 2. Instructions for Continued Airworthiness 0743-08471-2348 Revision Titled 24 September 2021; 3. Airplane Flight Manual Supplement 0643-08470-2348 Revision D dated 14 May 2020; 4. later revisions of the above list.</td>
<td>This STC is applicable to all serial numbers listed in the Delta Engineering Master Data List. Operational approval to use T-PED’s must be obtained separately from the STC. This STC includes activation of existing installed In-FlightEntertainment equipment installed in a configuration presented in Delta...</td>
<td>20/07/2022</td>
<td>Active</td>
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<td>STC Holder 10047020</td>
<td>TENENCIA Ltd.</td>
<td>ATR 7-212A</td>
<td>ATR 72-101/-102/-201/212 ATR 42-200/-300/-320/-400/-500</td>
<td>ATR - Installation of a Single UNS-1Lw Universal Avionics FlightManagement System with LPVThe purpose of this modification is to install a single UniversalAvionics Systems Corporation (UASC) UNS-1Lw Navigation Management System to comply with Precision RNAV and Approach Operations to LocaliserPrecision...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, Tenencia Ltd Report 713, Issue 5, dated 8 November 2013; 2013AFM supplement TEN/ ATR/005 (Precision RNAV withLPV Approach); issue 8, dated 11 November 2013; 2013AFM supplement TEN/ ATR/006 (Precision RNAV and RNP Approach); issue 7, dated 11 November 2013; Instructions for Continued Airworthiness...</td>
<td>This approval is limited to the aircraft complying with prerequisites/aircraft configuration defined in §6 of Master Data List, Tenencia Ltd Report 713, Issue 5; dated 8 November 2013. Prior to installation of this design change it must be determined that the interfactual relationship between this design change and any other previsously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft product.</td>
<td>12/11/2013</td>
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<td>10047026</td>
<td>CRANFIELD AEROSPACE LTD</td>
<td>Cessna 550 Citation</td>
<td>STM00005A - Navigation - Upgrade from Garmin 1000 installations to Garmin G1000/ G1200/ G1000H</td>
<td>The Certification Basis for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certification noise and emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>Cessna Aircraft Ltd Type Design Definition Amendment No. CAA/D001/14 dated 08 November 2013 or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the new installation introduces no adverse effect to the airworthiness of the product. The installation of this modification by third party</td>
<td>12/15/2013</td>
<td>Active</td>
<td>10047026</td>
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<td>10047057</td>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>208</td>
<td>Replacement of the original Pratt and Whitney PT6A-144(A) engine serial number UG-198600 and the 3-blade Hartzell Propeller HC-64N-P/D90000(K) and associated optional equipment in accordance with FAA STC SA02467LA.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): 23-ACE-3, Dynamic Evaluation, Engine Installation: Factors from this special condition were incorporated into the whirlmode flutter analysis per 14 C.F.R.</td>
<td>AFMS 2012011-1-Rh(-&gt;) FAA approved June 10, 2013</td>
<td>Prior to installation of this design change it must be determined if the new installation introduces no adverse effect to the airworthiness of the product. The installation of this modification by third party</td>
<td>14/11/2013</td>
<td>Active</td>
<td>10047057</td>
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<tr>
<td>10047061</td>
<td>SCANDINAVIAN AVIONICS DESIGNS A/S</td>
<td>Cessna 550 Citation</td>
<td>STM00005A - Universal UNS FMS System Evaluation for R-VNAV Approval</td>
<td>The Certification Basis for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certification noise and emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>Scandinavian Avionics, AFM Supplement 20120926-1, Issue 1, dated 09/2012 or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the new installation introduces no adverse effect to the airworthiness of the product. The installation of this modification by third party</td>
<td>15/11/2013</td>
<td>Active</td>
<td>10047061</td>
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<td>10047083</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>PA-24-250, PA-24-260</td>
<td>Installation of 3,04409 MT-Propeller Model MTV-9/8-188-50.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: FAR 23.1529 Amdt. 28, CS-36 Amdt.2 and FAR 38 Amdt. 28</td>
<td>According to MT-Propeller Master Document List No. E-2338 and MT-Propeller Installation Instructions Doc. No. E-2340 or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined if the new installation introduces no adverse effect to the airworthiness of the product. The installation of this modification by third party</td>
<td>14/11/2013</td>
<td>Active</td>
<td>10047083</td>
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<td>10047083</td>
<td>BAESYSTEMS (OPERATIONS) LIMITED</td>
<td>Cessna 550 Citation</td>
<td>STM00005A - Navigation - Upgrade from TCAAS II Version 7.0 to TCAAS II Version 7.1 on Boeing 757-200/ 757-200PF</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 12CS 25.1232, 25.1233, 25.1459. CS 25.1529, CS 25.1581.</td>
<td>BAESystems Major Change Dossier AW1742/B737 Addendum 1 Issue E-2339 to be maintained in accordance with EASA</td>
<td>Prior to installation of this design change it must be determined if the new installation introduces no adverse effect to the airworthiness of the product. The installation of this modification by third party</td>
<td>29/09/2014</td>
<td>Active</td>
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<td>10047084</td>
<td>BAESYSTEMS (OPERATIONS) LIMITED</td>
<td>Cessna 550 Citation</td>
<td>STM00005A - Navigation - Upgrade from Garmin 1000 installations to Garmin G1000/H1200/H1000H</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 12CS 25.1232, 25.1233, 25.1459. CS 25.1529, CS 25.1581.</td>
<td>BAESystems Major Change Dossier AW1742/B737 Addendum 2 Issue E-2339 to be maintained in accordance with EASA</td>
<td>Prior to installation of this design change it must be determined if the new installation introduces no adverse effect to the airworthiness of the product. The installation of this modification by third party</td>
<td>16/09/2015</td>
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<td>EASA STC no.</td>
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<td>Limitations/Conditions</td>
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<td>10047098</td>
<td>ST ENGINEERING AEROSPACE LTD.</td>
<td>Modification of Boeing 757-200CF Main Deck Cargo Door; Modification of the main deck cargo door to remove the cargo door limitation related to EASA STC 10038822</td>
<td>ST Aerospace MEL 3360000, Revision G. 12/11/2013 APM Supplement: 33/8A/TSE/AFM. Rev A. 11/11/2013 CA Supplement: 33/6AIVAC/AFM. Rev 0. 11/11/2013 APM Supplement: 33/6AIVAC/AFM. Rev 0. 12/11/2013</td>
<td>STC is limited to an aircraft number 25622 only incorporated this STC removes the 23rd December 2013 limitation related to the Cargo Door as imposed by EASA STC 10038822. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other...</td>
<td>18/11/2013</td>
<td>Active</td>
<td>10047098</td>
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<td>10047100</td>
<td>KUERZI AVIONICS AG</td>
<td>THRANE &amp; THRANE AVIATOR 300 SATCOM system WiFi option activation of F50EX</td>
<td>Master Document List doc no. MEL-201212903 Rev A dated 24 November 2013; Aircraft Flight Manual Supplement doc no. FMS-201212903 Rev 0 dated 20/12/12 or later revisions of the above listed documents approved by EASA.</td>
<td>Restricted to SIN 315, Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/11/2013</td>
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<td>10047100</td>
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<td>10047116</td>
<td>GAMA ENGINEERING LTD.</td>
<td>Installation of Garmin G600 PFD/ND according to Gama Engineering Ltd. modification no. - 1294</td>
<td>Master Documents List no. MDL-1294, issue 2 dated 08/10/2013; Flight Manual Supplement no. AFM-1294, issue 1 dated 05/04/2013; Installation Procedures no. IPI-1294, issue 1 dated 05/04/2013; Instruction s for Continued Airworthiness no. ICA-1294, issue 1 dated 08/04/2013; or later revisions of the above list...</td>
<td>This installation is not compatible with SOCATA modification no. MOD70-176-00 or MOD70-276-00 (Garmin G1000 flight deck installation). Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change...</td>
<td>19/11/2013</td>
<td>Active</td>
<td>10047116</td>
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<td>10047138</td>
<td>THUNDER BAY AVIATION LTD.</td>
<td>Elevator &amp; Ruddler Trim Tab Control Cable Replacement</td>
<td>Thunder Bay Aviation Ltd. Master Drawing List MDL1001 Rev. 0. TCCA approved June 4, 2013; or later revisions of the above listed documents approved by EASA.</td>
<td>Eligible control cables are listed on TCCA approved STC SA11-53 issue 1 or later TCCA approved version. The Thunder Bay Aviation Trim Tab Control Cable can be manufactured and installed in accordance with TCCA approved MDL 1001 (see above). Prior to installation of this design change it must be determined...</td>
<td>20/11/2013</td>
<td>Active</td>
<td>10047136</td>
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<td>10047152</td>
<td>AERO VODOCHODY AEROSPACE A.S.</td>
<td>Implementations of provisions for EFB class 2</td>
<td>Master Data List ref. AM007-46-20-MDL, issue b, dated 20/11/2013; Drawing List ref. AM007-46-20-DL, issue - dated 24/06/2013; Certification Plan ref. AM007-46-20-EC, issue - dated 24/06/2013; AFFM Supplement ref. AM007-46-20-AFM, issue -; date 24/06/2013; AFFM Supplement ref. AM007-46-30-AMM, issue -; date 24/06/2013; AFFM Supplement ref. AM007-46-30-AMM, issue -; date...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>22/11/2013</td>
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<td>List of EASA Supplemental Type Certificates</td>
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<td><strong>STC Holder, Model, Description, Certification basis</strong></td>
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<td>PW Inc., 300, B300C, LED Lighting for Cabin</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Installation of Light Emitting Diode (LED) lighting in cabin overhead. The installation of this design change must be determined through an analysis that demonstrates that the new design change does not affect the airworthiness of the original product.</td>
<td>Prior to installation of the design change it must be determined through an analysis that demonstrates that the new design change does not affect the airworthiness of the original product.</td>
<td>20/11/2013</td>
<td>Active</td>
<td>10447156</td>
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<td>H4 AEROSPACE (UK) Ltd., A320-214, Honeywell TCAS II system upgrade to change 7.1. Installation of Honeywell TPA-100B TCAS II Traffic Computer, p/n 940-0351-005, to existing aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Modification Summary Sheet HAA1673 Issue 2, dated 28/09/2017. This approval is limited to the installation of this design change in accordance with instructions for Continued Airworthiness. Prior to installation of this design change it must be determined through an analysis that demonstrates that the new design change does not affect the airworthiness of the original product.</td>
<td>None</td>
<td>05/12/2017</td>
<td>Active</td>
<td>10447157</td>
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<td>RUECKER AEROSPACE GmbH, A320-232, Installation of cabin, seat and electrical system modifications in support of the DLR SANDRA project.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Master Document List MDL-239-0007-000, dated 20/06/2013. This approval is limited to the installation of this design change in accordance with instructions for Continued Airworthiness. Prior to installation of this design change it must be determined through an analysis that demonstrates that the new design change does not affect the airworthiness of the original product.</td>
<td>None</td>
<td>22/11/2013</td>
<td>Active</td>
<td>10447182</td>
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<td>PMV ENGINEERING, 757-200, In-Flight Entertainment System Installation on Boeing 757-200. This STC changes the cabin layout (introduction of new seats and new DPA). In a new IFE System, developed by the IMS Company, embedded displays incorporated to the new seats: Notes: 1. This STC constitutes an AMOC to AD 2007-1...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-02 Heat release and smoke demands requirements to seat materials; The requirements for environmental protection and the associated certified noise...</td>
<td>Master Document List, ref.: SCA-00240-MDL-R01, dated 21 March 2014. This approval is limited to the installation of this design change in accordance with instructions for Continued Airworthiness. Prior to installation of this design change it must be determined through an analysis that demonstrates that the new design change does not affect the airworthiness of the original product.</td>
<td>1. This approval is limited to the installation of this design change in accordance with instructions for Continued Airworthiness. Prior to installation of this design change it must be determined through an analysis that demonstrates that the new design change does not affect the airworthiness of the original product.</td>
<td>24/03/2014</td>
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<td>10447189</td>
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<td>VECTOR AEROSPACE HELICOPTER SERVICES INC., AS 332 C, L, L1, Installation of an Astronics Luminous Lighting System in accordance with Vector Aerospace Helicopter Services USA Inc. Master Drawing List, HP 332-33-50-1028 Revision B, dated April 2013. This STC is the validation of FAA STC DR 2398 AK (Date amended August 22, 2012.</td>
<td>The Certification Basis (SB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- Vector Aerospace Helicopter Services USA Inc. Master Drawing List, HP 332-33-50-251 Revision A, dated April 23, 2013. This STC is the validation of FAA STC DR 2398 AK (Date amended August 22, 2013). - Vector Aerospace Helicopter Services USA Inc. Master Drawing List, HP 332-33-50-301 Revision A, dated April 23, 2013. This STC is the validation of FAA STC DR 2398 AK (Date amended August 22, 2013).</td>
<td>The installation of this modification is considered supplementary to existing emergency lighting system. Any rotorcraft incorporating this change must also be equipped with auxiliary and unmodified emergency lighting system conforming to the original Type Design. Prior to installation of this design change it must be determined through an analysis that demonstrates that the new design change does not affect the airworthiness of the original product.</td>
<td>25/11/2013</td>
<td>Active</td>
<td>10447200</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
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<th>Model</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>10047213</td>
<td>AMTECH AERONAUTICAL LIMITED</td>
<td>TCIA C-LSA11-017/D ISSUE 1</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Frequency Converter System Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC No. C-LSA11-017/D Issue 1 dated 14 January 2011</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>25/11/2013</td>
<td>Active</td>
<td>10047213</td>
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<tr>
<td>10047221</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA ST0011GSNY</td>
<td>A340-641, A340-642, A340-541, A340-542 A340-300</td>
<td>Installation of Panasonic Global Communication Suite (GCS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Delta Engineering Airplane Flight Manual Supplement No. ES727-108936-120 Rev 1 dated 10 September 2014</td>
<td>1. Operational approval to use T-PEO’s must be obtained separately from this STC. 2. T-PED’s is approved by EASA STC 10052464 (required configurations approving activation of eXConnect, eXW and/or eXPhone). As when installing/activating the GCS on the aircraft the aircraft must comply with...</td>
<td>24/11/2015</td>
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<tr>
<td>10047222</td>
<td>STANDARD AERO</td>
<td>FAA STC ST00784CH</td>
<td>727, 727C 727-L00/-200/-300</td>
<td>Installation of dual Universal-Avionic UMS-10 u2/UniversalAvionicUMS-F Flight Management Systems with optional Universal MAD-640 Rev. 1: RNP limitation removed.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative environmental requirements and its associated certification/approval.</td>
<td>Master Data List 1021803 Rev. F Airplane Flight Manual Supplement No. ES727-108936-100 Rev 2 dated 10 September 2014</td>
<td>As per AFMS Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>16/12/2016</td>
<td>Active</td>
<td>10047222</td>
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<tr>
<td>10047226</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>FAA STC ST00784CH</td>
<td>PA-46-500TP</td>
<td>Installation of 5-blade MT-Propeller ModelMTV-27-1-E-C-F-RpYCFR023-88d</td>
<td>- ICA Instructions for Continued Airworthiness Doc. No. E-2173 and MT-Propeller Installation Instructions Doc. No. E-2169 or later Revisions To be operated in accordance with MT-Propeller ICA Document No. E-2710 later revisions of the above,</td>
<td>According limitation section of AFM-SP. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>26/11/2013</td>
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<td>10047253</td>
<td>GFD GmbH</td>
<td>FAA STC ST00784CH</td>
<td>L31A 31A</td>
<td>FMIS/GPS Antenna Provision Rev. 1: Transfer to GFD GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- ICA Instructions for Continued Airworthiness Doc. No. 5290-075-Master Data List Doc. No. 5290-049r for later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
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<td>10047254</td>
<td>GFD GmbH</td>
<td>FAA STC ST00784CH</td>
<td>L31A 31A</td>
<td>VHF Antenna Provision Rev. 1: Transfer to GFD GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- ICA Instructions for Continued Airworthiness Doc. No. 5290-075-Master Data List Doc. No. 5290-049r for later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
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<td>GFD GmbH</td>
<td>10047255</td>
<td>REV. 1</td>
<td>10047325</td>
<td>L31A, 35A, 38A 31A, 35A, 35A</td>
<td>UHF Antenna Provision Rev. 1: Transfer to GFD GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- 16C Instrs for Continued Airworthiness Doc. No. 5260-066- Master data list Doc. No. 5260-048r later revisions of the above listed documents/approval/acceptance under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce roadnoise effect upon the airworthiness of the product.</td>
<td>23/06/2022</td>
<td>Active</td>
<td>10047255</td>
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<tr>
<td>ANDORA INTERNATIONAL LLC</td>
<td>10047256</td>
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<td>10047280</td>
<td>767-300</td>
<td>Installation of Mechanical Cargo Loading System Main Deck for Boeing Converted Freighters (BCF).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.29, 25.30, 25.30A, 25.30B, 25.30C, 25.30D, 25.47, 25.47I, 25.61, 25.611, 25.605, 25.606, 25.607, 25.609, 25.6...</td>
<td>Ancra International LLC Controlled Data List DLB6250. Revision 6. 30/09/2013 later revisions of the above listed documents/approval/acceptance under the EASA system.</td>
<td>To be installed on aircraft converted to a freighter by the BCF reconfiguration. Boeing 767-00-0039 Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce roadnoise effect upon the airworthiness of the product.</td>
<td>28/11/2013</td>
<td>Active</td>
<td>10047256</td>
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<td>AIRBUS S.A.S.</td>
<td>10047280</td>
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<td>10047325</td>
<td>A319-115</td>
<td>Airbus Corporate Jet Centre - Top Modification C21230 - VIP Project «LAGUNA BLANCA»</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The General CRI A-4301-013 «Aircraft Project Certification Basis» (appendix to this TV) The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification Approval Sheet, ref. C21230, Issue 1, dated 29/12/2013</td>
<td>This STC is limited to MSN 4768.</td>
<td>29/11/2013</td>
<td>Active</td>
<td>10047280</td>
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<td>CONCORDE BATTERY CORPORATION</td>
<td>10047325</td>
<td></td>
<td>10047280</td>
<td>EMB-605</td>
<td>Concrete Lead Acid Battery replacement for the existing Lead Acid Batteries on the Embraer EMB-505 «PHENOM 300»</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. Since the EASA certification basis for the original product is different from the FAA Certification basis, CRIF-01 &quot;Special condition for battery endurance requirements&quot; (which is part of the original...</td>
<td>Master drawing list DWG No S-0449 revision B dated 17 July 2012/13/2012. AFA revision is associated to this change. Later revisions of the above listed documents/approval/acceptance under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce roadnoise effect upon the airworthiness of the product.</td>
<td>03/12/2013</td>
<td>Active</td>
<td>10047325</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>10047327</td>
<td></td>
<td>10047327</td>
<td>S8 AND S6</td>
<td>Installation of S-TEC System 56/55X Two Axes Automatic Flight Guidance System, Model S7-650.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with S-TEC Bulletin No 650, Revision 1, dated 01-08-16 and Master Drawing List No S7600, Revision A, dated 3-18-16 (28 Volt System). Operation in accordance with FAA Approved Pilot’s Operating Handbook and/or Aircraft Flight Manual Supplement Part No 891919, dated 08-08-94 or FAA...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce roadnoise effect upon the airworthiness of the product.</td>
<td>03/12/2013</td>
<td>Active</td>
<td>10047327</td>
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<tr>
<td>STC Holder</td>
<td>Type</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>INDUSTRIAL NEOTEX S.A.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>SF 25 B</td>
<td></td>
<td>Use of Unleaded Avgas in SF25B with Starm MS1500 engine</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. To be installed in accordance with &quot;Certification Program for STC No.0010025506&quot; dated 02.12.2013. To be operated in accordance with AFMS in appendix 2 of &quot;Certification Program for STC No.0010025506&quot; dated 02.12.2013. To be maintained in accordance with ICA in appendix 3 of &quot;Certification Program for STC No.0010025506&quot; dated 02.12.2013.</td>
<td>See EASA STC for latest amendments. Prior to installation of this design change it must be determined that the interrelationship between this change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/12/2013</td>
<td>Active</td>
<td>10047351</td>
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<tr>
<td>TELEDYNE UK LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>767-400ER</td>
<td>767-200, 767-300, 767-300F, 767-200-300X-300F-400ER</td>
<td>Installation of Enhanced Airborne Data LoaderRev. 1: Transfer from Teledyne Limited to Teledyne UK Limited. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: JAR E25 Change 14: JAR 25.23, JAR 25.25, JAR 25.27, JAR 25.561, JAR 25.603, JAR 25.605, JAR 25.606, JAR 25.1301, JAR 25.1309...</td>
<td>Preparatory to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/11/2020</td>
<td>Active</td>
<td>10047365</td>
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<td>AIRGLAS, Inc.</td>
<td>FAA STC SD0094AK</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td></td>
<td>Fabrication and installation of the Airglas, Inc., Landes PA32-206LANose Fork equipped with Goodyear 8.50 x 6 tire in accordance with FAASTC 100-00047361 as last amended by FAA June 5, 2006 or later FAA approved. The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airlges Master Data List PA32-206LA, Revision D, dated May 26, 2006/Aviation Flight Manuals see AMLor later revisions of the above listed document(s) approved/accepted under the EASA system. 1. Airplanes incorporating this STC may be equipped with 6.00-6, 7.00-6, 8.50 x 6 tires on the nose and main wheels. The nose tire must be the same size as the main tires. All installations - 2. Compatibility of this STC with previously approved modifications must be determined by the installer.</td>
<td>05/12/2013</td>
<td>Active</td>
<td>10047387</td>
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<tr>
<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A321-213(1)/213(2)/A321-111(1)/213(2)/211(1)/211(2)/A320-216(+)/216(1)/216(2)/216(3)/A320-216(4)/214-215/A319-131-131/A319-131(1)/211/A319-131(1)/213/A319-131(1)/213(1)/131(2)/2111/213(1)/211(3)/A318-111(1)/212(1)/212(1)-122</td>
<td>Installation of HBL Power Systems Ltd P/N NCSP B 23006LM Ni-Cad Battery</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Compliance Report, ref. doc. TEN 625, Issue 2, dated 12th December 2013. Modification TEN 655, ref. doc. TEN 655, Issue 2, dated 4th December 2013. Declaration of Compliance for installation of HBL Power Systems Ltd at NCSP B 23060LM Ni-Cad Battery, dated 4th December 2013.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair are maintained in accordance with EASA's airworthiness documentation. The installation of this modification by third party.</td>
<td>05/12/2013</td>
<td>Active</td>
<td>10047388</td>
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<tr>
<td>BOMBARDIER INC.</td>
<td>TCCA STC SA07-5S</td>
<td>BD100-1A10</td>
<td></td>
<td>Installation of a Heads Up Technologies CMS-400 Digital Voice Check-List System in accordance with Modification Data Summary ES20200 Revision A dated September 12, 2012.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification Data Summary ES20200 Revision A dated September 12, 2012. - Airplane Flight Manual Supplement H63000 REV N/C dated 2 August 2012. - Instructions for Continued Airworthiness and supplemental maintenance Information H030700/CA REV N/C dated 10 August 2012.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair are maintained in accordance with EASA's airworthiness documentation. The installation of this modification by third party.</td>
<td>05/12/2013</td>
<td>Active</td>
<td>10047389</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>AVIATION SUPPORT GmbH</td>
<td>REV. 1</td>
<td>MD-900</td>
<td>Installation of Searchlight Trakkabeam, A380 on MD900/902. Rev 01. Removal of CAT-A limitations</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product ar unchanged and remain applicable to this certificate approval. Flight Manual Supplement FMS-AE-9901D rev 1-0, dated 30 October 2016, or later revisions of the above listed documents approved by EASA. Instructions for Continued Airworthiness ICA-AE-99019 Rev 1-0, dated 20 October 2018, Master Document List MDL-AE-91610, initial issue, dated 27 June 2017.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a negative effect upon the airworthiness of the product.</td>
<td>24/06/2017</td>
<td>Active</td>
<td>10047374</td>
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<tr>
<td>GULFSTREAM</td>
<td>REV. 2</td>
<td>FAAC STAC ST04291AT-D</td>
<td>GV-SP</td>
<td>Installation of an Executive Completion Interior</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CR-6-01 “Shower Installation” Equivalent Safety Finding(s): CR-6-50 “Cabin pressure and Saemorrhage”. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. FAA Approval ST04291AT-D, Date of Issuance 13th November 2013 - FAA Statement of Compliance 100G-13-275, dated 9th December 2013 - FAA Approved Index List GS52008877, Revision 1, dated 13 November 2013 - Lufthansa Flight Manual Supplement - AFMS - GC52020M142, Revision 1, dated 13 November 2013.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a negative effect upon the airworthiness of the product.</td>
<td>06/12/2013</td>
<td>Active</td>
<td>10047380</td>
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<td>JANCO AMERICA, INC.</td>
<td>REV. 1</td>
<td>FAA STAC FAA ST04291AT-D</td>
<td>MD-11F</td>
<td>Installation of a new courier module on MD-11F series airplanes.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. IA-11MDL{&lt;,&gt;}(=./) Revision B, December 27, 2005 - Master Data List (&lt;,&gt;)(=./)DC-085, Revision A, June 29, 2004. Certification Plan EFB module (&lt;,&gt;)(=./)S-4793, Revision NEW, October 24, 2004 - Instructions for Continued Airworthiness. FAA Approved Index List, dated 2013.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a negative effect upon the airworthiness of the product.</td>
<td>09/12/2013</td>
<td>Active</td>
<td>10047398</td>
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<tr>
<td>P3 ENGINEERING GMBH</td>
<td>REV. 2</td>
<td>B737-800</td>
<td>Installation of Interioor Attachment Components</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: C/S 25 amendment 12 for the affected area. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Instructions for Continued Airworthiness PV/100018-IA-0001-Maintenance Manual Supplement PV/100018-MMS-0001-M Maintenance Planning Document PV/100018-MPD-0001-Maintenance Weight and Balance Report PV/100018-VBMS-0001-M Master Document List PV/100018-MDL-0001-A, issued December 09, 2013.</td>
<td>Maximum Loads at interface points of attachments not to be exceeded for follow-on aircraft modifications as per PV/100018-MA-0009-D. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change is not to exceed the limits for aircraft modifications.</td>
<td>09/12/2013</td>
<td>Active</td>
<td>10047402</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>777F</td>
<td>Installation and activation of an Electronic Flight Bag (EFB) on ab777-TFCs STC installations (using provisions from OEM options) and activates aClass 2 EFB in the cockpit, composed of touch screen displays mounted at the OEM locations for the LH and RH Class 3 EFB, and docking station/transition personalization.</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: C/S 25 at Amend 3 for the following paragraphs: 20-13027 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Engineering Bulletin, Lufthansa Technik 777- EB46-0069 revision 02, Engineering Bulletin, Lufthansa Technik 777- EB46-0073 revision 03, Aeronautical Flight Manual Supplement, Lufthansa Technik A-06-1554- AFMIssue 03 revision 03, Instructions for Continued Airworthiness, Lufthansa Technik A-06-1354-ICA Issue 0.</td>
<td>1. The following Boeing options must be installed on aircraft prior to subject modification: 4615E161A11, 4615E161A12, 4615E161A20-2. The following Engineering Bulletins, approved as minor changes under Lufthansa Technik D0A, are required to be performed prior to the subject modification:</td>
<td>22/10/2014</td>
<td>Active</td>
<td>10047408</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>STC Holder</td>
<td>767-200</td>
<td>Modification to Existing Interface Completion</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Transport Canada STC No. C-LSA13-104D Issue 1 dated 31 October 2013/ACS-MAI Ltd., Modification Data Summary J08000 Revision NC</td>
<td>For both systems: List of Documents: N16-EC145 HV2+ NUM, Rev. E-01 For the HELICOM V2 NUM system: Certification Programme N2-EC145 HV2+ NUM-E-A-04 dated 4th December 2013/Installation Manual N7-EC145 HV2+ NUM-E-A-04 dated 4th December 2013/Maintenance Manual Supplement; N6-EC145 HV2+ NUM-E-A-04 dated 4th December 2013</td>
<td>03/07/2024</td>
<td>Active</td>
<td>10047431</td>
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<tr>
<td>SAFRAN ELECTRONICS &amp; DEFENSES/SIS UNIT - HELICOPTERS DATA MANAGE MT SOLUTIONS &amp; SERVICES</td>
<td>REV. 10</td>
<td>MBB-BK117 D-3M MBB-BK117 D-3 MBB-BK117 D-2 MBB-BK117 C-2</td>
<td>Installation of Flight Data Monitoring and Transmitting System HELICOM V2 NUM or HELICOM V2+ NUM/initial. The HELICOM V2 NUM system consists of a recording/ transmittal offlight and system's parameters/revision 1: this change extends the capabilities of the system by introducing PN 4723P / PN 4733P/ The est...</td>
<td>Flight Manual Supplement E101530, IR, TCCA approved February 9, 2012/RMA Modification Instruction E101510, Rev. A (aeroplane)/RMA Modification Instruction E101511, IR (engine) or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EASA/ The installation of this modification by third persons is subject to written permission of the approval holder/ The HELICOM V2 NUM system is approved for the model MBB-BK117 C-2/ The HELICOM V2+ NUM system is approved for the models MBB-BK117 D-2/ MBB-BK117 D-3 and MBB-BK117 D-3m/ Prior to...</td>
<td>24/03/2022</td>
<td>Active</td>
<td>10047437</td>
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<tr>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>TCAC STC SA12-13</td>
<td>DHC-6 SERIES 100 DHC-6 SERIES 100, SERIES 200/ Installation of Electronics International MVP-6L Engine Display System/ Kawasaki TCCA STC SA12-13</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CR1 F-01 &quot;Lihtium battery installations&quot; dated 11th December 2013 (raised in EASA project No. 001002654#5-001 - technical STC) for the est...</td>
<td>Flight Manual Supplement E101530, IR, TCCA approved February 9, 2012/RMA Modification Instruction E101510, Rev. A (aeroplane)/RMA Modification Instruction E101511, IR (engine) or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EASA/ This installation is approved for the DHC-6 airplane carrying 8 passengers or less. For DHC-6 airplane carrying 10 or more passengers/Electronics International MVP-6L Standby Torque indicator(s) must be installed to display the torque information of both engines/ Prior to installation of this design change...</td>
<td>12/12/2013</td>
<td>Active</td>
<td>10047440</td>
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<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 1</td>
<td>767-200</td>
<td>767 Comlux Aviation Cabin Modifications are defined in the &quot;Project description/ Doc. No. 767 Comlux Project Description/ Rev 00/ Master Data List/ Doc. No. S.OAVED.00003- MDL, R00Rev. 1/ Transfer of STC from 767 Comlux Aviation Systems AG to...&quot;</td>
<td>- Master Data List/ Doc. No. S.OAVED.00003- MDL, R00AFMS/ Doc. No. S.OAVED.00003- AFMS Rev 00/ ICA/ Doc. No. S.OAVED.00003- ICA Rev 00/ Prior to installation of this design change...</td>
<td>20/04/2021</td>
<td>Active</td>
<td>10047441</td>
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<td>DASSAULT FALCON SERVICE</td>
<td>FALCON 10</td>
<td>Universal Avionics MFD-640 upgrade on Falcon 10</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>DFS Change Approval Sheet doc. DFS1300-CAS Issue A dated 06December/2013/DFS Modification Sheet doc. DFS1300-CAS Issue A dated 06 December 2013 or later revisions of the above listed documents approved by EASA/ Prior to installation of this design change...</td>
<td>13/12/2013</td>
<td>Active</td>
<td>10047436</td>
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<td>EASA STC No</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC Ref</td>
<td>Model</td>
<td>Description</td>
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<td>10047457</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 2</td>
<td>737-400, 737-800, 737-600, 737-700, 737-800,</td>
<td>Smart POCIA Card (SPC) installation for the SAGEM &quot;Wireless Extension For ACMS&quot; system. The Certification Basis for the original product as amended thêm future additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: CS25 and 4.25.601, 25.603, 25.611, 25.869, 25.899, 25.1301(b)-(c), 25.1307(b), 25.1309, 25.1353, 25.1357, 25.1367, 25.1369, 25.1373.</td>
<td>Certification basis: Master Data List 0550-01-A2103-R051P. This Certification Basis is applicable to this certificate and/ or repair. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/10/2014</td>
<td>Active</td>
<td>10047457</td>
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<tr>
<td>10047459</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>FALCON 10</td>
<td>FMS UNS-1L Activation of Approach Mode on Falcon 10</td>
<td>The Certification Basis for the original product as amended thêm future additional or alternative airworthiness requirements: CS 25 Amdt 3: §1302(c) The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification basis: DFS Change Approval Sheet doc. DFS1299-CAS Issue A dated 06 December 2013. DFS Airplane Flight Manual Supplement N° F10-038A. Original issue date: 12 December 2013. or later revisions of the above listed documents approved by EASA.</td>
<td>FMS airworthiness compliance is demonstrated for: RNP APCh/with Baro-VNAV (AMC 20-27)- Conventional approaches (GPS, VOR, VOR/ DME) (AC 20-133A) GPS overlay approaches (AMC 20-5) Approaches are authorized only in manual and without flight director. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/12/2013</td>
<td>Active</td>
<td>10047459</td>
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<td>10047463</td>
<td>MEGGOT SAFETY SYSTEMS, Inc.</td>
<td>FAA SA3439NM</td>
<td>727-200F</td>
<td>Installation of Systron Donner Smoke Detectors Part Number 7740 on Boeing 727-200F Series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification basis: System Donor Control Drawing 7940, Revision &quot;B&quot;, dated October 26, 1987. Component Maintenance Manual 26-14-66 rev 7 dated 2012-07-15 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 200404/CF (or subsequent revisions of this decision) and/or the...</td>
<td>A mixed configuration of this installation and that of the existing smoke detector installation is not allowed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will not introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/12/2013</td>
<td>Active</td>
<td>10047463</td>
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<td>10047481</td>
<td>LUFFTHANSA TECHNIK AG</td>
<td>A320-321/341-322/343</td>
<td>EBF ConnectivityConnection of installed EBF Class 2 to company server via aircraftdatalink (ATSU communication function)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification basis: Compliance Record List CRL64-122M, Iss./Rev: 2.1, Dated 4th of December 2013. or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third...</td>
<td>16/12/2013</td>
<td>Active</td>
<td>10047481</td>
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<td>10047490</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>BD-700-1A10</td>
<td>Boeing 700 SATCOM System Installation in Global Express BD-700</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification basis: GEX01-581001 Service Bulletin EDGEX01-HDN02 Flight Manual Supplement EDGEX01-HDN02 Maintenance Manual Supplement EDGEX01-HDN02 Maintenance revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third...</td>
<td>16/12/2013</td>
<td>Active</td>
<td>10047490</td>
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<tr>
<td>10047491</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 2</td>
<td>A340-311/312/313</td>
<td>eXPhone System Activation on A340-300 with eXConnect. Description: - Summary: The Panasonic eXPhone system allows passengers to use their own cellphones or &quot;TIPED&quot; (i.e., &quot;Transmitting Personal Electronic Devices&quot;) while in flight for cellular voice and Public services GSM (voice and text) and GPRS/EDGE (data) services. - Prior to installation of this modification it must be determined that the interrelationship between this...</td>
<td>Certification basis: Master Document List (MDL): ref 0634-02-A2103 rev 2- Instructions for Continued Airworthiness. This Certification Basis is applicable to this certificate and/ or repair. Prior to installation of this modification it must be determined that the...</td>
<td>Prior to installation of this modification it must be determined that the...</td>
<td>11/05/2016</td>
<td>Active</td>
<td>10047491</td>
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<td>EASA STC No</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10047494</td>
<td>BRITISH AIRWAYS FLOBRITISH AIRWAYS ENGINEERING FICE OF AIRWORTHINESS</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>A340-313, A340-642</td>
<td>Airbus TCAS 7.1 UPDATETh е n н е р w i t h T h e a s a e at 000-050 to correct False Raisaue</td>
<td>The Certification Basis for the original product as amended by the following modification: at a later amendment; CS25 amendment 13 for environmental protection and the associated Ideas and emissions levels of the original product unchanged and accurate to this certificate/approval.</td>
<td>- Master Data List - Doc ref: BA-340-34-0001-MDL Issue 3- STC Definition &amp; Certification Document - Doc Ref: BA-340-34-0001-STC Issue 3- Compliance Check List - Doc Ref: BA-340-34-0001-CLL Version 000 later revisions of the above listed documents approved by EASA. Service Bulletin No. BA-340-34.</td>
<td>Prior to installation of this design change must be determined by the aircraft manufacturer.</td>
<td>06/11/2017</td>
<td>Active</td>
<td>10047494</td>
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<td>10047498</td>
<td>PW ENGINEERING GmbH</td>
<td>REV. 1</td>
<td>ATR 42-500</td>
<td></td>
<td>Integration of a Flight Inspection System into an ATR 42-500 The purpose of this modification is to integrate a Flight Inspection System from Aerodata into a VIP configuration ATR 42-500. Thermal installation consists of additional sets of antennas, an Observer/Console in the cabin facing a standard double window.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental Aircraft Flight Manual (AFM), PP ref. AT45A-AMF-A revision 01 dated 21 November 2013.</td>
<td>27/05/2014</td>
<td>Active</td>
<td>10047498</td>
<td></td>
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<td>10047501</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 1</td>
<td>B200</td>
<td>MOD 13126 - WiFi and Immersion Installation</td>
<td>CS-23 Amdt. 3 for electrical, avionic and cabin modification requirements as listed on document AE13126 PCI rev. 3. The requirements for environmental protection and the associated certified noise and emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List AE13126MDL Rev dated 11/12/2013 Flight Manual Supplement AE13126AFMS Rev dated 13/08/2013, Instructions for Continued Airworthiness AE13126IC Rev dated 02/07/2013, Rear Tail Tray Installation AE13126UC Rev dated 16/07/2013, Safety Precracy for Ferry Flight AE13126UCG Rev dated 11/12/2013.</td>
<td>Limitation to serial number BB-1006 Prior to installation of this change it must be determined by the aircraft manufacturer. The installation of this change and any other previously installed design change and/or repair will introduce aerodynamic effect upon the airworthiness of the product. Prior to installation of this design change it must be determined by the aircraft manufacturer. The installation of this change and any other previously installed design change and/or repair will introduce aerodynamic effect upon the airworthiness of the product.</td>
<td>17/12/2013</td>
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<td>10047524</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>737-800/900-900ER 737-300-400-500-600-700</td>
<td>TCAS System Upgrade to Change 7.1 on Boeing 737 Aircraft for Honeywell Systems</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>- Modification Summary Sheet MOD HA-1675 Issue 2-Airplane Flight Manual Supplements HAFAFSMS34 Issue IR, HAFAMS30 Issue/for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this change it must be determined by the aircraft manufacturer. The installation of this change and any other previously installed design change and/or repair will introduce aerodynamic effect upon the airworthiness of the product. Prior to installation of this change it must be determined by the aircraft manufacturer.</td>
<td>22/04/2014</td>
<td>Active</td>
<td>10047524</td>
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<td>10047525</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 1</td>
<td>MF20 GF</td>
<td></td>
<td>Mode EHS &amp; ADS-B deselection The aim of the change is to have ADS-B capabilities approved as per EASA AMC 20-24, and comprises the following set of modifications, as per the Certification Program reference: AE13120PDCP Rev. 3, dated 16/12/2013. Installation of two GPS antennas, PIN 581-517-145. Installed:</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List ref. AE13120MDL Rev. 1 dated 16/12/2011, later revisions of the above listed documents approved by EASA.</td>
<td>Applicability limited to Syphire Falcon 20 GF model, SN 397-. The current ADS-B out installation has been certified in accordance with the EASA AMC 20-24 for Non-Recorder areas, with the related limitations stated in the AFMS ref. AE13120AFMS Rev. 0. Additionally, the installation is limited to specific aircraft as per AFMS ref. AE13120AFMS Revs.</td>
<td>19/12/2013</td>
<td>Active</td>
<td>10047525</td>
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<tr>
<td>10047526</td>
<td>AERODATA AG</td>
<td>B300</td>
<td></td>
<td></td>
<td>Installation of Aerodata Flight Inspection System AD-ACS-113</td>
<td>The Certification Basis for the original product remains applicable. Limitations not affected by the modification, for items affected by the modification, the following new Certification Basis is elected to comply: CS-23 Amdt. 3, except 23.561 and 23.785, FAR 23.561 Amdt. 34 and FAR 23.785 Amdt. 3. Installation in accordance with the Master Data Document. Aerodata Doc No. MDL-A/5-488. Rev. 2 dated 04 December 2013, Aircraft Flight Manual Supplement, Aerodata Doc No. AFM-139, Rev. 0 dated 26 November 2013, Aircraft Maintenance Manual Supplement, Aerodata Doc No. AM-1144, Rev. 0 dated 03 December 2012.</td>
<td>Only applicable for Soral PL-69/2013 Prior to installation of this change design change it must be determined by the aircraft manufacturer. The installation of this change design change and any other previously installed design change and/or repair will introduce aerodynamic effect upon the airworthiness of the product.</td>
<td>18/12/2013</td>
<td>Active</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10047564</td>
<td>HELI-ONE (NORWAY) AS</td>
<td>1</td>
<td>AS332L/AS332L.1</td>
<td>MOD AHSE-64046 VOR Receiver upgrade.</td>
<td>The Certification Basis for the original product as amended by retrofitting additional or alternative airworthiness requirements.</td>
<td>CS 29 Amendment 3 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain as...</td>
<td>- Hel- One AHSE-64048 General Arrangement. Hel- One AHSE-64048 Rev A Embodiment instructions. Hel- One AHSE-64048 Reg A Certification Plan.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/12/2013</td>
<td>Active</td>
<td>10047564</td>
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<td>10047589</td>
<td>MT-PROPPELLER ENTWICKLUNG GmbH</td>
<td>2</td>
<td>PA-30, PA-39 (TWIN COMANCHE)</td>
<td>Installation of 3-blade MT-Propeller Model MTV-12-CC-PCF(L)185+B7 coPiper PA-30, PA-39 (Twin Comanche).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. This certificate/approval involves a change to the requirements for: environmental protection or a change to the certified noise or emissions levels.</td>
<td>According to MT-Propeller Master Document List No. E-2535 and MT-Propeller Installation Instructions Doc. No. E-2535, orders/Revisions to be operated in accordance with MT-Propeller AFM/S Document No. E-2349 to be maintained i.a.w. MT-Propeller ICA Document No. E-2351 or later revisions of the above...</td>
<td>According limitation section of AFM/SP. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/12/2013</td>
<td>Active</td>
<td>10047588</td>
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<tr>
<td>10047608</td>
<td>HELI-ONE (NORWAY) AS</td>
<td>1</td>
<td>AS332L, AS332L.1</td>
<td>MOD AHSE - 65865 &lt;(&gt;,&lt;&gt;) EURONAV 7 Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval.</td>
<td>CS 29 Amendment 3 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain as...</td>
<td>- Hel- One Flight Manual Supplement NYH(No-201 &quot;EURONAV 7 INSTALLATION&quot;HS-636 83 Rev- dated 21 Dec. 2013 or later revisions of the above listed documents approved by EASA. - Hel- One AHSE-65865 Rev B Embodiment instructions. Hel- One AHSE-65865 Rev A.</td>
<td>Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/12/2013</td>
<td>Active</td>
<td>10047608</td>
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<td>10047619</td>
<td>ACS-NAI Ltd.</td>
<td>2</td>
<td>CL-600-2B19</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 8103</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval.</td>
<td>Transport Canada STC No. C-LSA10-047/D Issue 1 dated 18 October 2015/CS-NAI Ltd. Modification Data Summary G425000 Revision NC.Aero Consulting Services Ltd. Mandatory Maintenance Manual. Supplement G192070 Revision NC.Aero Consulting Services Ltd. Mandatory Maintenance Manual. Supplement G90971 Revision...</td>
<td>Restricted to ASN 8103 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/01/2014</td>
<td>Active</td>
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<td>10047621</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 2</td>
<td>SA 228T-(B)</td>
<td>Installation of Sandel SN 4500 EHSI and L3 Avionics SKY 897AS.</td>
<td>The Certification Basis for the original product as amended by retrofitting additional or alternative airworthiness requirements. The following paragraphs(s) is a later amendment: CS 23 Amnt. 2, § 867, 1301, 1309, 1311, 1321, 1322, 1327, 1351, 1353, 1357, 1365, 1387, 1431, 1529, 1541, 1551 The requirements...</td>
<td>Master Data List AE12064566VL Rev. 4 dated 21 December 2015/Flight Manual Supplement AE12064566VMG45530AW Rev. 1 dated 18.12.2013/Flight Manual Supplement AE12064566SH/SH0998 Rev. 0 dated 18.07.2012/Flight Manual Supplement AE12064566S/SH/SH0998 Rev. 0 dated 17.05.2013 or later revisions of the above listed documents...</td>
<td>This installation is approved for IFR B-NAV, P-NAV, RNP FM Doppler, LNAV and RVSM operations, with coupling to AP 80 or M4D autopilot RNP LMV - VNAV- APV bars approaches are not authorized. The installation of this modification by third persons is subject to written permission of the approval holder. Prior to...</td>
<td>25/07/2018</td>
<td>Active</td>
<td>10047621</td>
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10047652
STC Holder: SAFRAN POWER USA, L.L.C.
FAA STC0156CH
60
Certification basis:
Fitment of Safran starter generator by STC Installation of Safran starter-generator module 23091-009 and QAD kit 23091-500 on Learjet 60.
The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.
Goodrich drawing(s):<br>23091-009 rev. C, dated August 9, 2015 Goodrich drawing(s):<br><br>Associated Technical Documentation:
Revision of referenced ICA documents approved by EASA inconcurrence with Goodrich.<br>Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.
09/01/2014 Active 10407020

10047653
STC Holder: AIRBORNE TECHNOLOGIES GmbH
REV. 2
P.68TC "OBSERVER" P.68R "VICTOR"
P.68 OBSERVER 2 P.68C, P.68C-TC
P.68 Mission cabin configuration with Option for Flight into Known icing for P.68R "VICTOR" STC-AT-071(1) certifies the Mission cabin configuration, prequalified the general installation into the P.68. The STC-AT-071(1) consists of the following steps which can be performed as required: Step 1: "Mission cabin configuration, P.68 Mission cabin configuration with "OBSERVER" 737-800 LUFTHANSA"
"Collectors" VIP Cabin Completion Rev. 1: Update of referenced ICA documentation.
Change/Repair Approval, Doc. No. STC-071C-G-AS, Rev. Installation Instructions, Doc. No. 071-II-STC-AT, Rev. 1, dated 02/03/2015 (related to Step 1) to Installation Instructions, Doc. No. STC-071C-G-AS, Rev. 0, dated 08/09/2019 (related only to Step R1) for Step 1 of STC-AT-071-AM Supplement, Doc. No...
The Installation of this STC-AT-071(1) is only allowed in aircraft where Letter of Authorization "STC-071(1)-G-LOA" has been issued by AirbornTechnologies GmbH. Prerequisites for each individual step of this STC are defined in section 1.1 of the installation instructions. Doc. No. 071-d-STC-AT.
20/09/2019 Active 10407603

10047657
STC Holder: LUFTHANSA TECHNIC AG
REV. 1
737-800 "Collectors" VIP Cabin Completion Rev. 1: Update of referenced ICA documentation.
Supplemental Type Definition Document STCD00-1219M issue 03AFMS - 00-11/17TS-AMF issue SDCA A-00-11/17S-ICA Issue 04 Cabin Crew Manual A-00-11/17S-CAM issue 05 or later revision of the above listed documents approved by EASA.
The installation of STC 10047400 rev 0 is a prerequisite to this design change considered in the above mentioned documentation. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change...
22/07/2015 Active 10407607

10047658
STC Holder: TAIKOO (XIAMEI) AIRCRAFT ENGINEERING CO. LTD. xiaobei HAEBO XIAMEI
REV. 1
A319-115 VIP interior reconfiguration
The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s); - D-01 Installation of an open cockpit in the galley area of a VIP interior, D-05 Isolated Compartments, D-06 Glass Screens of Displays / Monitors...
Supplemental Type Definition Document STCD00-1219M issue 03AFMS - 00-11/17TS-AMF issue SDCA A-00-11/17S-ICA Issue 04 Cabin Crew Manual A-00-11/17S-CAM issue 05 or later revision of the above listed documents approved by EASA.
The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s); CRI-D-4302-011 Isolated Compartment / CRI-D-4306-007 "Seat Heat Release and Smoke Density", CRI-D-4310-011 Glass Screen of Display Monitors/ Deviation(s); - HKCAD STC-2013-12, Rev. 3, dated 10/03/2015, Master Drawing List TA19-AD-MOL Rev.AF No. 18/10/2013 (for MSN0523), Master Drawing List TA19-AD-MOL Rev.H No. 31/07/2014 (for MSN0588 and 5907), Compliance Summary Report TA19-AD-CMR Rev. D, dated 06/2015, Instructions for Continued Airworthiness. This approval is valid for installation on A319-115 MSN 5023, 5088 and 5907. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the... This approval is valid for installation on A319-115 MSN 5023, 5088 and 5907. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft. 17/11/2015 Active 10407609

10047659
STC Holder: WIPARE inc.
FAA STC02967CH
1. The installer must determine whether this design change is compatible with previous approved modifications. 2. If the holder agrees to permit another person to use this certificate to alter the product, the holder must give the other person written evidence of that permission.
09/01/2014 Active 10407609
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<th>FAA STC ref.</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>10047665</td>
<td>HARTZELL PROPELLER INC.</td>
<td>REV. 1</td>
<td>FAA SA03786C</td>
<td>3) 182C, 182T, T1827 (2) FR182 1) R182, TR182, T182</td>
<td>Installation of the Iridium Satellite System</td>
<td>FAA STC No ST04139AT</td>
<td>If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person writen evidence of that permission.</td>
<td>13/01/2014</td>
<td>Active</td>
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<td>10047686</td>
<td>STANDARD AERO</td>
<td>REV. 1</td>
<td>FAA STC ST0491AT</td>
<td>BD-700/1A10 BD-700/1A10</td>
<td>Installation of the TrueNorth Simphone</td>
<td>FAA STC No ST0491AT dated 08 February 2013</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of this product.</td>
<td>10/01/2014</td>
<td>Active</td>
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<td>10047688</td>
<td>PMV ENGINEERING</td>
<td>REV. 1</td>
<td>FAA STC ST0481AT</td>
<td>737-800</td>
<td>Activation of a Panasonic eXPhone system.</td>
<td>Certification Investigation Report SCA-00111-Rev 03/03/2016</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of this product.</td>
<td>16/12/2015</td>
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<td>10047708</td>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST04140AT</td>
<td>767-200-300/300F</td>
<td>Installation of the Rockwell Collins Aircraft Communication, Addressing and Reporting System (ACARS), including Cockpit Voice Recording (CVR) Updates Revision 1 transferred from EMEQ Engineering to B/E Aerospace.</td>
<td>Master Drawing List: 2-130380-021, Revision A, 04/13/2013</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of this product.</td>
<td>02/03/2016</td>
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<td>10047709</td>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST04139AT</td>
<td>767-200-300/300F</td>
<td>Installation of the Iridium Satellite Communication (SATCOM) System Revision 1 transferred from EMEQ Engineering to B/E Aerospace.</td>
<td>Master Drawing List: 2-130385-001, Revision B, 03/10/2013</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of this product.</td>
<td>02/03/2016</td>
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<td>STC Holder</td>
<td>GNS 530 upgrade to WAAS Unit incl.</td>
<td>AB139, AW139</td>
<td>Insulator Installation on Helicopter AW30 (Change 139P005-00)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-RFM Supplement Doc. 139P050-00-RFS-00</td>
<td>Issue of later revisions of the above listed documents approved by EASA.</td>
<td>15/02/2014</td>
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<td>STC Holder</td>
<td>A321-213/-231/-232</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBILITY</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification Summary Sheet H4A0659 Issue 17, dated 08/10/2007</td>
<td>Instructions for Continued Airworthiness H4AD0778 Issue 10, dated 18/09/2017</td>
<td>Prior to installation of this modification it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/10/2017</td>
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<td>STC Holder</td>
<td>PA-32-301T</td>
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<td>GNS 530 upgrade to WAAS Unit incl.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAR 23Amit 21: The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List: 1268-17, dated 20.11.2013</td>
<td>Prior to installation of this modification it must be determined that the relationship between this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/04/2014</td>
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<td>STC Holder</td>
<td>AS355NP</td>
<td></td>
<td>Flight with Cabin Vibration Damper</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:CS 27.1, 251., 301., 303., 305., 307., 601., 603., 771., 85., 1321., 1505., 1525., 1529., 1541., 1551., 1583.A27.1., A27.2., A27.3., A27.4 aer CS 27 Amit 37R The requirements listed in...</td>
<td>Flight Manual Supplement AS355 NP FMS079-002 Issue 1 dated 05 Nov 13 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/01/2014</td>
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<td>STC Holder</td>
<td>CAP 231, CAP 231 EX, CAP 232</td>
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<td>PROPELLER MT14V1 BC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/01/2014</td>
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<td>STC Holder</td>
<td>FAA STC ST01199CH-D</td>
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<td>Validation of FAA STC ST01199CH-D Installation of AlliedSignal Global Airborne Flight Information System(AFIS) Satlink System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/01/2014</td>
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<td>JCB AERO SA</td>
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<td>10047885</td>
<td>A320-214</td>
<td>A320 cabin reconfiguration from 158 to 180 pax.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Compliance Report A320-00883-01-CR-01 Rev. JR, dated 17/01/2014 and Master Drawing List A320-00883-01-MDL-01 Rev. C, dated 19/01/2014</td>
<td>Prior to installation of this design change it must be determined that it will not introduce noise/shake effect upon the airworthiness of the product.</td>
<td>22/01/2014</td>
<td>Active</td>
<td>10047799</td>
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<tr>
<td>PW ENGINEERING GmbH</td>
<td></td>
<td>10047829</td>
<td>767-400ER</td>
<td>Structural provisions for VIP Cabin Completion and for the installation of: (&gt;) quad belly camera(&gt;) - Iridium sat antenna(&gt;) - Door locks(&gt;) - avionic rack.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Drawing List B767A-MDL-50100 Issue B dated 16.01.2014 and Certification Document List CCN-B767A Issue 01 dated 20.01.2014</td>
<td>Before flight, entry and access doors must comply with the prerequisite in the AFM supplement B767A-AMF-A. Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will not introduce noise/shake effect upon the airworthiness of the product.</td>
<td>23/01/2014</td>
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<td>AIRGLAS, Inc.</td>
<td></td>
<td>10047850</td>
<td>R44, R44i</td>
<td>Fabrication and installation of composite floorboard scuff plates Fabrication and installation of of Airglas, Inc. LF2600-R44 composite floorboard scuff plates in place of original carpet floor covering.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Descriptive Data List LF-2600-R44, revision (-) dated July 10, 2013 and Descriptive Data List LF-2600-R44, revision (-) dated July 8, 2013</td>
<td>Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will not introduce noise/shake effect upon the airworthiness of the product.</td>
<td>24/01/2014</td>
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<td>EUROCOPTER CANADA Ltd.</td>
<td></td>
<td>10047881</td>
<td>2 A 8 355, E/F/F1/F2/N/P 1</td>
<td>Rear Cargo Door Hinge and Latch Modification</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Eurocopter Canada Ltd Master Drawing List MDL-98-008 Issue 00 of Eurocopter Canada Ltd Instructions for Continued Airworthiness CA-EC1-207, rev. 2c</td>
<td>Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will not introduce noise/shake effect upon the airworthiness of the product.</td>
<td>27/01/2014</td>
<td>Active</td>
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<td>CESSNA DUESSELDORF CITATIONSERVICE CENTER GmbH</td>
<td></td>
<td>10047885</td>
<td>CESSNA 441</td>
<td>Installation of KH1-1050 System</td>
<td>The Certification Basis for the original product as amended by the following: An additional or alternative airworthiness requirements: Amdt. 23-20: 23-130/Amdt. 23-21: 23-101b Amdt. 23-23: 23-603, 23-606 Amdt. 23-26: 23-159/Amdt. 23-45: 23-57 (a),(c) Amdt. 23-45: 23-305, 23-615, 23-773/Amdt. 23-46: 23-301</td>
<td>AFMS Report 2013-9-8, Rev. A, dated 25 November 2013</td>
<td>Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will not introduce noise/shake effect upon the airworthiness of the product. The installation of this modification by third p.</td>
<td>27/01/2014</td>
<td>Active</td>
<td>10047885</td>
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List of EASA Supplemental Type Certificates

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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
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<th>Foreign/NA</th>
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<tr>
<td>10047886</td>
<td>MID-CANADA, MOD CENTER</td>
<td>REV 1</td>
<td>TCCA SA12-39</td>
<td>ATR 72-202</td>
<td>ATR 42-300, ATR 42-320</td>
<td>Dual ULS-1ue FMS Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Master Drawing List: Avionics Design Services Ltd. MDA12008; Revision B, Issue 1, Transport Canada approved May 22, 2014; For ATR 42 FMS Supplement, Avionics Design Services Ltd. FMS12008-1; Revision A, Transport Canada approved May 15, 2014.</td>
<td>Compliance with Airworthiness Limitations, Avionics Design Services Ltd MDA12008-1 for ATR 42 or MDA12008-2 for ATR 42, Revision NF C, Transport Canada approved April 26, 2012 (or later EASA approved revision).</td>
<td>30/06/2016</td>
<td>Active</td>
<td>10047886</td>
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<tr>
<td>10047894</td>
<td>PRECISE FLIGHT, Inc.</td>
<td>REV 1</td>
<td>FAA STC TO0453SE</td>
<td>B737-800, -900ER</td>
<td>B737-400/500/600/700, B737-100-200-200C-300</td>
<td>Installation of Pulselight landing light flashing unit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Instructions for Continued Airworthiness 152PMAN0318 Rev C dated May 23, 2012/see attached table/for subsequent revisions of the above listed documents approved by EASA.</td>
<td>as per AFMS NIGHT OPERATION LIMITATION Turn Pulselight System Off at 500 AGL or at Pilots Discretion During Landing Approach (if TCAS/TCAD option is installed), also Turn Pulselight/Strobe Light Switch To OFF.</td>
<td>28/01/2014</td>
<td>Active</td>
<td>10047894</td>
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<tr>
<td>10047909</td>
<td>ALPINE AEROTECH LP</td>
<td>REV 1</td>
<td>TCCA STC SH01-32 ISSUE 1</td>
<td>212, 412, 412EP</td>
<td>Installation of Alpine Aerotech Ltd. Exhaust Duct Assembly</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Master Drawing List 191M-99-004-MSL; Rev. NC, dated 11February 2000. Installation Instructions 191M-99-004-MMS, Rev. NC, dated 11February 2000.</td>
<td>This approval is limited to Bell helicopter models 212, 412, and 412EP (the model 412 CF being bilateral agreement), is approved by EASA in accordance with EASA DO-254/201104-01, Revision NC dated 14 January 2019.</td>
<td>10/02/2014</td>
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<td>10047916</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV 1</td>
<td>CAA UK RESPONSIBIL</td>
<td>737-700</td>
<td>Introduction of Emergency Locator Transmitter</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>STC Twenty One Limited Master Drawing List S21.MDL-057 Initial Issue dated 17/05/2013/STC Twenty One Limited Maintenance Manual S21.TEC-035, Issue 1, dated 16/11/2013</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair.</td>
<td>28/01/2014</td>
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<td>10047926</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV 1</td>
<td>757-200</td>
<td>Activation of IMA Rave IFE system</td>
<td>Rev 1; Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>AES List Master Data List No. AES-757-2003 MDL Issue 5, dated 28th January 2014</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair.</td>
<td>27/04/2021</td>
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<td>EASA STC No.</td>
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<td>Limitations/Conditions</td>
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<td>10047937</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 1</td>
<td></td>
<td>A321-213/-211/-212</td>
<td>Installation of Provisions for Class 2 Electronic Flight Bags, Rev. 1: Transfer from Teranics Ltd. to Aerosmile and A320-111 removed from applicability list.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification of TEN 734, Airbus A320 Series, Installation of Provisions for Class 2 Electronic Flight Bags, Issue 1 dated 14 January 2014 Flight Manual Supplement TEN.</td>
<td>Prior to installation of this design change/repair it must be determined that the interrelationship between this design change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/06/2022</td>
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<td>10047945</td>
<td>WIPARE Inc.</td>
<td>FAA STC NO. SA1185CH</td>
<td>B) U206G, TU206G</td>
<td>A) 206H, T206H</td>
<td>Installation of Wipaire Model 3450 Amphibian/Seaplane Floats and Hartzell Propeller Model HC-F3YR-1RF/ F486A-2R (see under limitations).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>A) Flight Manuals Models 206H and T206H: Pilot's Operating Handbook and FAA Approved/Airplane Flight Manual Supplement, dated 25 February 2000; Models U206G and TU206G: Airplane Flight Manual/Pilot's Operating Handbook Supplement, dated August 21, 2000 with Soley Turbine Pac and Sloey Mark II. B) 1...Installation of Wipaire Model 3450 Amphibian/Seaplane Floats and Hartzell Propeller Model HC-F3YR-1RF/F486A-2R only for Cessna 206H and T206H.</td>
<td>This approval is limited to only the installation made in Gulfstream Model GV-SP Serial No. S2601. It is the installer responsibility to ensure that all STCs fitted are EASA approved. Prior to installation of this design change it must be determined that the interrelationship between this design change and...</td>
<td>03/01/2014</td>
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<tr>
<td>10047981</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST03770AT-D</td>
<td>GV-SP</td>
<td>Installation of a 16-passenger cabin interior with one additional crew seat</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements (or equivalent Safety Finding(s)) CRI D-50 Issue 1 Cabin Egress and Seat Encroachment. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.</td>
<td>-FAA STC ST03770AT-D, dated 5th May 2010- Gulfstream Aerospace Corporation Index List GCS200080M, Revision H, dated 5th May 2010 GV-SP Airplane Flight Manual Supplement GCS20009117 and GCS20009119, original release, dated 5th May 2010. -Instructions for Continued Airworthiness GCS20001400, original rev.</td>
<td>This approval is limited to only the installation made in Gulfstream Model GV-SP Serial No. S2601. It is the installer responsibility to ensure that all STCs fitted are EASA approved. Prior to installation of this design change it must be determined that the interrelationship between this design change and...</td>
<td>03/02/2014</td>
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<td>10048007</td>
<td>BRITISH AIRWAYS PLOEBRITISH AIRWAYS ENGINEERING OFFICE OF AIRWORTHINESS</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING 757-200</td>
<td>B757-200 TCAS change 7.1 update</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements C5 Amendment 13:CS 25.891, CS 25.896, CS 25.896(1&lt;-&lt;)&gt; C5 25.1301, CS 25.1301, CS25.1301(1-&lt;&gt;)&lt; C5 25.1322, CS 25.1351, CS 25.1353, C5 25.1357, CS25.1431, CS 25.1.</td>
<td>Master Data List Ba-757-34-0007-MDL, Issue 0/1, dated 2010, later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/02/2014</td>
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<td>10048008</td>
<td>AVIATION COMMUNICATION &amp; SURVEILLANCE SYSTEMS, L.L.C. (Acss)</td>
<td>FAA STC ST03770AT-D</td>
<td>727 SERIES</td>
<td>Installation of Honeywell Traffic Alert and Collision Avoidance System TCAS II change 7.1 software</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Index of Master Data Lists 9500000 rev D/Airplane Flight Manual Supplement TCAS-APPS-0294 rev. C9902400 0/1, dated 2004, later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/02/2014</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>AIRCRAFT TECHNICAL DEVELOPMENT INC dba AEROCON ENGINEERING COMPANY</td>
<td>STC02473LA</td>
<td></td>
<td>DHC-8-102/103/106</td>
<td>Installation of Cabin Interior Noise Reduction Kit</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (a) CS 25.811 Am1d 5, (b) CS 25.853 Am1d 12, (c) CS 25.866 Am1d 8, and (d) CS 25, App H25 Am1d 14</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended must be determined and remain applicable to this certificate/ approval.</td>
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<td>06/02/2014</td>
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<td>FIELD AVIATION EAST, LTD.</td>
<td>REV. 1</td>
<td>TCZA SA13-64</td>
<td>DHC-8-301-311-314-315 DHC-8-102-103-106-201-202</td>
<td>Dual Universal RCU-4100, Radio Control Unit Retrofit</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended remain applicable to this certificate/ approval.</td>
<td>Installation must be in accordance with FAA approved Aircraft/Engineering Company Master Drawing List AC25- NRDHC8-1000, Revision B. Maintenance must be in accordance with MaintenanceManual Supplement/MSM- DHC8-1805, Revision B or later revisions of the above listed documents approved by EASA in accordance...</td>
<td>The following equipment must be installed as a prerequisite to the installation of the Universal RCU-4100 system: Rockwell Collins communication and navigation radio.- A Field Aviation or Bombardier OEM installation of Dual UNS-1(U) Flight Management System with operating software SCN 802090X or later...</td>
<td>12/02/2014</td>
<td>Active</td>
<td>10048022</td>
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<td>ROEDER PRAEZSION GMBH</td>
<td>REV. 4</td>
<td>SA227-OC, SA227-AT, SA227-TT, SA227-AC, SA227-BC, SA226-AT, SA226-T, SA226-T(B)</td>
<td>Installation of TCAS II System and Mode S Transponder with upgrade of Thales Pascal CAS-67A to SIV– Change 7 - Rev 2. Transfer of STC Rev 3: Combination of STC 10048031 with STC No. 2004-5755 Rev 4: Type correction to STC 10048031 AFM supplement reference.</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as unchanged must remain applicable to this certificate/ approval.</td>
<td>As listed in the AFM. Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td></td>
<td>01/03/2023</td>
<td>Active</td>
<td>10048031</td>
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<td>BRITISH AIRWAYS AIRWAYS PL/B/R/S AIRWAYS ENGINEERING OFFICE OF AIRWORTHINESS</td>
<td>A321-211 A320-211/214 A319-111</td>
<td>A321-211</td>
<td>Installation of TCAS II System and Mode S Transponder with upgrade of Thales Pascal CAS-67A to SIV– Change 7 - Rev 2. Transfer of STC Rev 3: Combination of STC 10048031 with STC No. 2004-5755 Rev 4: Type correction to STC 10048031 AFM supplement reference.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (a) CS 25.811 Am1d 13 for the paragraphs affected by the change/ The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended remain applicable to this certificate/ approval.</td>
<td>Master Data List BA-320-34-0007-MDL, Issue 00, dated 10/10/2013 Compliance Check List BA-320-34-0007-CCL, Issue 00, dated 06/10/2013 STC Definition &amp; Certification Document BA-320-34-0007-STC, Issue 00, dated 10/10/2013 or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2014</td>
<td>Active</td>
<td>10048034</td>
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<td>S-TEC CORPORATION</td>
<td>FAA SA0252AC-D</td>
<td>R172E, R172F, R172G, R172H</td>
<td>Installation of S-TEC System 20/30 Single and Twin Axis Automatic Flight Guidance Systems, Model ST-606-2030 (4 Volt System)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended must be determined and remain applicable to this certificate/ approval.</td>
<td>FAA/DAS Approved Supplemental Flight Manual, PIN 891395, dated 23 May 1997 for Cessna R172E, R172F, R172G, and R172H for S-TEC System 20; FAA/DAS Approved Supplemental Flight Manual, PIN 891396, dated 23 May 1997 for Cessna T182, T182C(-()&lt;), T175B and PC-175C for S-TEC System 20; FAA/DAS Approved Supplem...</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2014</td>
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<td>10048043</td>
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List of EASA Supplemental Type Certificates
easa.europa.eu

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<th>STC Holder</th>
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<tr>
<td>GFD GmbH</td>
<td>REV. 2</td>
<td>L31A, E35A, E36A, 31A, 35A, 36A</td>
<td>IFF Mode 1 System Installation (ST/700) Rev. 2: Transfer to GFD GmbH</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certified noise/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data Doc. No. 5308-049 Rev A of 30.01.2014/ Instructions for Cont. Airworthiness Doc. No.5308-085 Rev IR of 09.12.2013/PM Supplement Doc. No. 5306-071 Rev IR of 04.12.2013/lor revisions of the above listed document(s) approved/ accepted by the EASA system.</td>
<td>Limited to S/N: 31A-205 &amp; 31A-207, 35A-139, -179, -268, -331, -367, 36A-062, -069(r-v) &gt; 35A-222, 35A-607, 35A-612, 35A-843 and 35A-387 Prior to installation of this modification it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>23/06/2022</td>
<td>Active</td>
<td>10048049</td>
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<tr>
<td>GYRO- STABILIZED SYSTEMS, LLC</td>
<td>REV. 4</td>
<td>FAA STC SR02420AK</td>
<td>R44, R44 II</td>
<td>ZattWorks R44M-001 Utility Camera Mount. The ZattWorks R44M-001 Utility Camera Mount is a forward mount for an existing front facing camera to the R44/R44II helicopter. Revision 1: administrative review Revision 2: model R44 added, update of document references Revision 3: transfer of EASA STC 10048064 to different STC ref.</td>
<td>Zattworks document number R44M-008, Revision 3, dated February 2015</td>
<td>25/10/2022</td>
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<td>NETJETS AVIATION, INC.</td>
<td>REV. 1</td>
<td>FAA STC SAD1727WH</td>
<td>EMB-505</td>
<td>Interior Alteration for installation of a Keurig Coffee Maker and modification of “No Smoking” signs in Embraer Model EMB-505 aircraft. The STC revision is due to the transfer of FAA STC SAD1727WH to a new certificate holder.</td>
<td>Installation in accordance with Duncan Aviation Inc. Master documentation No EDDOC-000015626, Rev A, dated 28 August 2013/Operation in accordance with Duncan Aviation Inc Airplane Flight Manual Supplement No EDDOC-000015927, Rev A, dated 28 August 2013</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>21/06/2022</td>
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<td>EASA STC no.</td>
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<tr>
<td>10048070</td>
<td>BELL</td>
<td></td>
<td>FAA STC SR02818NY</td>
<td>BELL 212</td>
<td>Bed Helicopter Vibration Monitoring System (BHVM) - The Bell Helicopter/Vibration Monitoring Kit (BHVM) is a system that allows to monitor the airflow, rotor and drive system vibration. The BHVM is designated to allow the operator to make vibration related track and balance adjustment between operations...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unachieved and remain applicable to this certificate/ approval.</td>
<td>RFMS A-0033 Rev. 02 dated 08.12.2009/CA AA-0853 Rev. 5 dated 06.05.2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/EASA Bilateral Agreement.</td>
<td>10/02/2014</td>
<td>Active</td>
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<td>10048074</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td></td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>B757-200</td>
<td>Change of Wing Fuel Tank Pump Relays to DFI type on Boeing 737-200/200ER, 737-700/200ER, 757-200ER/200ER.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:CS25.981 Amendment 1 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unachieved and remain applicable to this certificate/ approval.</td>
<td>Modification Summary: H4 Aerospace ref. MOD H4A/1690 issue 5, dated 06/February 2014.Instructions for Continuous Airworthiness. H4 Aerospace ref. H4ADR798-issue 1, dated 15 October 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>10/02/2014</td>
<td>Active</td>
<td>10480704</td>
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<tr>
<td>10048090</td>
<td>JCB AERO SA</td>
<td></td>
<td>A330-200</td>
<td>B/C and Y/C Seat Covers Installation Specification - Summary - Removal of existing B/C and Y/C Seat Covers - Installation of seat covers on the seats.</td>
<td>The certification Basis of the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unachieved and remain applicable to this certificate/ approval.</td>
<td>Project Data List: A332-00825-01-POL-01, rev. C Certification Programme 332-00825-01-CP-01, issue DICA, 332-00825-01-ICA-01, issue A130 revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this modification it must be determined that the installation of this modification does not adversely affect the environmental protection and the original product remains unchanged.</td>
<td>12/02/2014</td>
<td>Active</td>
<td>10480904</td>
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<td>10048104</td>
<td>AEROTEC &amp; CONCEPT</td>
<td></td>
<td>1900D</td>
<td>Approval of Beach 1900D for LPV Approach Capability.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unachieved and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Aeronet Modification ref AE13180 MDL, Rev 1, dated 04/December 2014. Operation in accordance with Aeronet Flight Manual Supplement ref AE13180AFMS, Rev 0, dated February 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>EASA STC ref 10034237 must be installed. Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2014</td>
<td>Active</td>
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<td>10048111</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td></td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>757-256</td>
<td>218Y/C, 757-200 passenger cabin reconfiguration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unachieved and remain applicable to this certificate/ approval.</td>
<td>H4 Aerospace Modification Summary and Declaration of Compliance No. MOD49A/1634 Issue 2 dated 12 February 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>13/02/2014</td>
<td>Active</td>
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<tr>
<td>10048125</td>
<td>AVIATION DEFENSE SERVICE</td>
<td></td>
<td>FAN JET FALCON</td>
<td>Carriage of PIRATE pods The purpose of this Change is to clear the carriage of PIRATE pods on Falcon 20 in the following envelope: VMO: 350 kts, MMO = 0.76; FL350; +0/-3g.</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval: CS 25 Amrt 11 §25.140CS 25 Amrt 10 §25.631FA 25 up to Amrt 43 §25.1323, §25.1325, §25.1520 The certified noise and/or emissions levels...</td>
<td>Change Approval Sheet doc. No. CAS-AVD- MDA20-95-FOG8A Issue 02 dated 12.02.2014 Master Document List doc. No. MDA-FVG-PDGA Issue 2 dated 12.02.2014 Aircraft Flight Manual Supplement. No. SAFM-AVD-FOG Rev. 01 or later revisions of the above listed documents approved by EASA.</td>
<td>Restricted to Falcon 20 S/N 076 aircraft. As per 414 Supplement N° 01/SAFM-AVD-FOG Rev. 01 Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>14/02/2014</td>
<td>Active</td>
<td>10481250</td>
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The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Installation of PFD Aspen EFD1000H Pro

The Certification Basis for the original product as amended by the following equipment: One Vertical Speed Indicator - One Vertical Speed Indicator.

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Installation of EC-280B in A320-200 for Etihad / Serbia

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

The Certification Basis for the original product as amended by the following equipment: One Flight Management System - One Flight Management System.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 13: CS 25.1302; CS 25.1309; CS 25.1312; CS 25.1313; CS 25.1314; CS 25.1459; CS 25.1529; CS 25.151.

Air Berlin Modification No. A320-EO25-2-01-2014 is a prerequisite. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 13: CS 25.1302; CS 25.1309; CS 25.1312; CS 25.1313; CS 25.1314; CS 25.1459; CS 25.1529; CS 25.151.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 13: CS 25.1302; CS 25.1309; CS 25.1312; CS 25.1313; CS 25.1314; CS 25.1459; CS 25.1529; CS 25.151.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 13: CS 25.1302; CS 25.1309; CS 25.1312; CS 25.1313; CS 25.1314; CS 25.1459; CS 25.1529; CS 25.151.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 13: CS 25.1302; CS 25.1309; CS 25.1312; CS 25.1313; CS 25.1314; CS 25.1459; CS 25.1529; CS 25.151.
### List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>URBE AERO GmbH</td>
<td>DA42, DA42 M</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>SBAS / The Gamin G1000 integrated flight deck is upgraded with SBAS capable equipment and the related operational capabilities are enabled.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td></td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the aircraft and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>24/02/2014</td>
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<td>1004207</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd</td>
<td>CAA UK RESPONSIBILITY</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>Installation of Cabin LED Lighting System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
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<td>20/02/2014</td>
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<td>SAA SOLUTIONS FOR AVIATION, S.L.</td>
<td>BD100-1A10</td>
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<td>Installation of Side Facing Divan in Challenger 300 aircraft</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CR D-01 Installation of Side Facing Seats</td>
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<td>None</td>
<td>20/02/2014</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd</td>
<td>CAA UK RESPONSIBILITY</td>
<td>REV. 1</td>
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<td>Installation of 228 Passenger LOPA</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
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<td>21/02/2014</td>
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<tr>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>LEARJET 45</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>LJ45 Special Equipment / Baggage Configuration LJ5</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): D-01, Cargo Fire</td>
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<td>05/09/2014</td>
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<td>J &amp; C ENGINEERING SERVICES Ltd.</td>
<td>737-700</td>
<td></td>
<td></td>
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<td>Passenger Seating Layout 50 (44 B/C + 6 Y/C) PAIX Change from 149 PAIX Layout to a 50 PAIX Layout</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td></td>
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<td>10048231</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST01812WI</td>
<td>ATR 72-211/212/212A ATR 72-101/102-201/202 ATR 42-200/300-320</td>
<td>ATR - Installation of a dual Garmin GTN-725 GPS Navigator System for European R-RNAV Compliance The purpose of this modification is to install a dual Garmin GTN-725 GPS Navigation system with 2 Garmin GA-36 WAAS/GPS antennas, as a replacement to the existing Honeywell GPS/GPS system (KL900 or KL900B).</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment:CIS25 and 4 for the following requirements:25.771, 25.775, 25.777, 25.1301, 25.1302, 25.1309, 25.1322, 25.1381, 25.1523, 25.1524, 25.1525, 25.1526, 25.1527</td>
<td>1. Master Document List STC Twenty One ref. S21 MFL-124 issue 1 dated 21 February 2014 or later revision approved by EASA.</td>
<td>02/02/2016</td>
<td>Active</td>
<td>10048231</td>
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<td>10048232</td>
<td>2 EXCEL DESIGN LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>PA-31</td>
<td>Introduction of modified winglets.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Installation in accordance with STC21 Modification S21.57-90-1001, S21 MFL-1010, Issue 1 dated February 21, 2014.</td>
<td>25/08/2017</td>
<td>Active</td>
<td>10048232</td>
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<td>10048260</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST03936AT-D</td>
<td>G150</td>
<td>Installation of Safe Flight Autopower Automatic Thrust Control System in accordance with Gulfstream Aerospace Corporation Index list CEC14348000 Revision K dated September 02, 2013, or later FAA approved revision.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CS 25.1329 g(b)(d) g(h)(k)(m) at Ammo 47% the requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Operation must be in accordance with FAA approved Flight Manual Supplement, CEC14348000, Revision B, dated March 26, 2013 Installation must be in accordance with Gulfstream Aerospace Corporation Index list CEC14348000 Revision K dated September 02, 2013. Maintenance must be in accordance with instructions.</td>
<td>- CATII operations with the ATS engaged are prohibited. Steep Approach Operations are prohibited below 400 ft AGL (or published minima, whichever is higher) with ATS engaged. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other...</td>
<td>25/02/2014</td>
<td>Active</td>
<td>10048260</td>
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<tr>
<td>10048265</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B737-300</td>
<td>Carriage of Horses or Other Animals</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CRI D 01 Fire protection of essential systems/ equipment within class E cargo compartments, CRI D 02 Installation of seats on B737-300 converted freighter. The requirement...</td>
<td>STC Twenty One Limited Flight Manual Supplement S21 FMS-0102 Issue 2 or later revisions of the above listed documents approved by EASA STC Twenty One Limited Master Document List No. S21 MFL-1133 Issue 2</td>
<td>The Following EASA STCs must have been previously embodied or embedded in a document with this design change. EASA STC 10015542 Rev 4 Flight Weight Conversion...</td>
<td>25/04/2016</td>
<td>Active</td>
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<td>10048272</td>
<td>ASI INNOVATION</td>
<td>REV. 1</td>
<td>SAA365 N</td>
<td>EHSIL and GPS installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement ASI-APM-0114-06 (date February 2014) or later revisions of the above listed documents approved by EASA</td>
<td>- The use of GPS for instrument approaches is not authorized A Minimum Equipment List must be established for B-RNAV and P-RNAV. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design changes...</td>
<td>26/02/2014</td>
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<td>10048272</td>
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<td>STC Holder</td>
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<td>Date of Issue</td>
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<td>PMV ENGINEERING</td>
<td>REV. 3</td>
<td>A321-213/231-232 A321-111/112-131-211-212 A322-231/252-232 A330-211/212-214/215-216 A318-151/162-130-131 A319-111/112-113-114-115</td>
<td>Installation of a Dual Advisory Ice Detection System on Airbus A320Rev initial System InstallationRev 1. System activationRev 2. Update of the applicabilityRev 3. Update of the MMEL.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged, remain applicable to this certificate/ approval without any...</td>
<td>Master Document List AVI-0035/MDL-R05 dated 8 June 2015 Aeronautical Flight Manual Supplement AVI-0035/AFMS-R01 dated 8 June 2018 Instructions for Continued Airworthiness AVI-0035/CA-R03 dated 13 March 2014 Master Minimum Equipment List AVI-0035/MMELS-S01-R00 dated 25 Feb. 2020 Master Document later rev...</td>
<td>Prior to installation of this change/repair it must be determined that the interference relationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/04/2020</td>
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<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC</td>
<td>727-100-200</td>
<td>Installation of New Flight Crew Oxygen Masks and holders.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Drawing List 98F9586-D01 rev. 0 FIlStructural Drawing List 98F9586-D02 rev AElectrical Drawing List 98F9586-D03 rev IRISystems Drawing List 98F9586-D04 rev Bor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2000/40/C (or subsequent revisions if...)</td>
<td>Prior to installation of this design change it must be determined that the interference relationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/02/2014</td>
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<td>B/E AEROSPACE LIMITED</td>
<td>REV. 2</td>
<td>TCCA SAI1-26, ISSUE 3</td>
<td>DHC-8-402</td>
<td>Upgrade Tail Navigation Light System: The Rev. 2 was issued due to the company's name change from &quot;ACS-NAI Ltd.&quot; to &quot;B/E Aerospace Ltd.&quot;</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation must be in accordance with TCCA approved ACS-NAI Ltd. Modification Data Summary designated G 855072, Revision A &amp; Illu...</td>
<td>Prior to installation of this design change it must be determined that the interference relationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/03/2017</td>
<td>Active</td>
<td>10048311</td>
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<tr>
<td>URBE AERO GmbH</td>
<td>REV. 1</td>
<td>DA40D</td>
<td>SBAS / The Gamin G1000 integrated flight deck is upgraded with SBAS-compatible equipment and the related operational capabilities are enabled.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>The change shall be accomplished in accordance with the initial revision of Doc. D037: Project No.: 018 and subsequent documents. The aircraft shall be operated in accordance with initial revision d05FM-018-01, and The aircraft shall be maintained in accordance with initial revision d05FM-018-01 for...</td>
<td>Prior to installation of this design change it must be determined that the interference relationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/02/2014</td>
<td>Active</td>
<td>10048316</td>
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<tr>
<td>RUAG AG</td>
<td>REV. 4</td>
<td>PC-12, PC-12/45, PC-12/47</td>
<td>Provisions to Aerial Survey Camera flight manual PC-12/47E. Doc. No.: FMS-11702, Rev. A. Flight Manual Supplement PC-12, PC-12/45 and PC-12/47. Doc. No.: FMS-60349, Rev. C. Instructions for Continuous Airworthiness PC-12/47E. Doc. No.:CA-800361 Rev E-Instructions for Continuous Airworthiness PC-12, PC-12/45, PC-12/47D...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any...</td>
<td>-Flight Manual Supplement PC-12/47E. Doc. No.: FMS-11702, Rev. A. -Flight Manual Supplement PC-12, PC-12/45 and PC-12/47. Doc. No.: FMS-60349, Rev. C. -Instructions for Continuous Airworthiness PC-12/47E. Doc. No.:CA-800361 Rev E-Instructions for Continuous Airworthiness PC-12, PC-12/45, PC-12/47D...</td>
<td>Prior to installation of this change/repair it must be determined that the interference relationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/01/2021</td>
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<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>1</td>
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<td>DC.8-72</td>
<td>Introduction of Swift Broadband/Installation of Falcom System, including wireless network for occupants for use of portable electronic device and internal access during flight. This Rev. 1 was issued to correct a typing error.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s): SCI3RIH01 EUW/G-CA. The following paragraph(s) at a later amendment: See Cert Program Doc. No. 521 - PROG-0249. This has also been applied on 757-200, 757-200P, 757-300.</td>
<td>Master Document List Doc. No. 521-MDL-0506(V)-1P-1, Flight Manual Supplement Manual Doc. No. 521-TEC-031, Compliance Report 521-COMP-069, Electrical Load Analysis 521-ELE-0131, Mass &amp; Balance Report 521-TMNP-0589r.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce any adverse effect upon the airworthiness of the product.</td>
<td>03/02/2014</td>
<td>Active</td>
<td>10448337</td>
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<tr>
<td>TELEDYNE UK LIMITED</td>
<td>1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>757-200, 757-200P, 757-300</td>
<td>Installation of Enhanced Airborne Data Loader Rev. 1. Transfer from Teledyne Limited to Teledyne Limited to Teledyne UK Limited.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - the following paragraph(s) at a later amendment: Note the EASA certification basis for the 757-300 is JAR 25 Change 14 but it has also been applied on 757-200, 757-200P.</td>
<td>Master Data List - MOD- MDL-031 Issue 01(+) or (+) later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce any adverse effect upon the airworthiness of the product.</td>
<td>26/11/2020</td>
<td>Active</td>
<td>10448344</td>
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<tr>
<td>ANORA INTERNATIONAL LLC</td>
<td>1</td>
<td>FAA STC SA553MN</td>
<td>727-200</td>
<td>Installation of a mechanical cargo loading system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Controlled Drawing List DL 81200, Revision &quot;A&quot;, dated September 8, 1988; Component Maintenance Manual Part Number 91200 rev 1 dated January 27th, 2014; for later revisions of the above listed documents approved by EASA inadvertantly.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce any adverse effect upon the airworthiness of the product.</td>
<td>05/03/2014</td>
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<tr>
<td>DART AEROSPACE USA, Inc.</td>
<td>1</td>
<td>FAA SR02233SE</td>
<td>412, 412EP, 205A-1, 212</td>
<td>Installation of 1st Pitch in Command KIT in accordance with dataprovided in Master Document List MDL-D212-803 Rev E as applicable to each helicopter model</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement FMS- D212-803 Rev C, 23rd February 2015 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement.</td>
<td>EASA STC is issued just for the helicopter Types/Models currently approved by EASA.</td>
<td>26/06/2016</td>
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<td>10448356</td>
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<tr>
<td>BRISTOW HELICOPTERS Ltd.</td>
<td>1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>AS 332 L1B1</td>
<td>Installation of a Co-Pilot alternative altimeter in accordance with Bristol Modification Document MOD-332/3376, Issue 6.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Bristol Modification Document MOD-332/3376, Issue 6. - Bristol Certification Plan BHL/332.7190 Issue A - Bristol Instructions for Continued Airworthiness BHL/332.7191 Issue A</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce any adverse effect upon the airworthiness of the product.</td>
<td>06/03/2014</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>1</td>
<td></td>
<td>650 CITATION VII</td>
<td>Honeywell FMZ-2000 Flight Management System P-NAV approval for existing Honeywell FMZ-2000This Rev. 1 was issued to correct a mistake in the name and address of the STC holder.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Scandinavian Avionics AFM Supplement, 2084533-AM, Issue 2, dated 28th August 2015; Scandinavian Avionics Installation Documents, 2084533 STC, Issue 1, dated 13 February 2014 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce any adverse effect upon the airworthiness of the product.</td>
<td>10/03/2014</td>
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<td>10448364</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<td>10048380</td>
<td>AVINCI S.A.</td>
<td>Installation of a new Universal Aircraft Interface Device (UAID) and Display Mounts</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Delta Engineering, Flight Manual, document No. 0688-0844-2338, original release, dated 10 September 2013. Delta Engineering, Master Data List, document No. 0188-0844-2338, revision A, dated 10 September 2013. Delta Engineering, Instructions for Continued Airworthiness and Electrical Wiring fit...</td>
<td>This STC is for provisions of Electronic Flight Bags and operational repair or maintenance of Portable Electronic Flight Bag (portable electronic device) including its battery that is not approved by this STC. This STC only includes approval for the Universal Aircraft Interface Device (UAID) for read only with existing avionic...</td>
<td>06/02/2014</td>
<td>Active</td>
<td>10448031</td>
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<tr>
<td>10048381</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>Installation of vortex generators on the wings and tail surfaces.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Micro Aerodynamics Drawing Package MA2134, Revision A, dated October 27, 2008, and Installation Manual MA2135, Revision A, dated October 28, 2008.</td>
<td>If more than five vortex generators are damaged or missing, the aircraft is not airworthy. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a total adverse effect upon the airworthiness of the product. The installation of this modification by third parties...</td>
<td>10/02/2014</td>
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<td>10448032</td>
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<tr>
<td>10048382</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>Project Specific Certification Plan for B777 Cargo Fire Extinguishers</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List 51751952, Revision B, dated 12 January 2010. Instructions for Continued Airworthiness 51751952, Revision C, dated 13 February 2010.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a total adverse effect upon the airworthiness of the product.</td>
<td>07/03/2014</td>
<td>Active</td>
<td>10448033</td>
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<td>10048383</td>
<td>MEGGITT SAFETY SYSTEMS, Inc.</td>
<td>Lighting System Upgrade</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Change/Repair Approval, Issuance document No. CR-1051-08, Change A and/or revision A, dated 5 February 2014. Continued Airworthiness Instructions, Issuance document No. CA-1051-08, Change A and/or revision A, dated 5 March 2014.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a total adverse effect upon the airworthiness of the product.</td>
<td>07/03/2014</td>
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<td>10048391</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAAD-STD SA082466</td>
<td>PA-28-160-180, PA-28-180-200, PA-28-160-180/180/235</td>
<td>Installation of vortex generators on the wings and tail surfaces</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation data with Drawing Package MA2150, Revision IR, dated August 28, 2000, and Installation Manual MA2151, Revision IR, dated October 2, 2000. Instructions for Continued Airworthiness, dated on Planard MA800 for later revisions of the above listed documents approved by EASA.</td>
<td>If more than five vortex generators are damaged or missing, the aircraft may not be airworthy. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/03/2014</td>
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<td>10048395</td>
<td>DASSAULT AIRCRAFT SERVICES (DASS)</td>
<td>REV. 4</td>
<td>FAA ST110475C</td>
<td>FALCON 2000, 2000EX</td>
<td>Installation of upgraded hardware to replace Rockwell Collins Proline 21 displays, add an Integrated Flight Information System (IFIS) and enable localizer performance with vertical guidance (LPV) capability, Rev. 4: Transfer from BHE &amp; Associated, Ltd. to Dassault...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product.</td>
<td>- Master Drawings List 365-00-001 Rev. C dated 07/03/2014, or later, approved FAA revision - Propeller Flight Manual Supplement, document No. 365-00-0056, Rev IR, dated 07/03/2014, for FAR 23.907, Amdt. 23-13-09, FAA approved February 11, 2014.</td>
<td>For aircraft with no flight data recorder (FDR), or FDR with less than 42 parameters, software version 9.0 or later is required. For aircraft with FDR with 42 parameters to 88 parameters, software version 10.4 or later is required. The installation requires the installation of EASA STC 10064620 and/or EAS...</td>
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<td>10048396</td>
<td>HARTZEIT PROPELLER INC.</td>
<td>FAA STC SA2014CH</td>
<td>TBM700A, B, C1, C2, N</td>
<td>Installation of Harzett 5-blade composite propeller HC-ESN-3CMC8834K</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements FAR 23.907, Amdt. 23-13. This certificate/approval involves a change to the requirements for environmental protection and the associated certified noise and/or emissions levels of the product.</td>
<td>Installation Instruction INST-0236, FAA approved February 11, 2014.</td>
<td>Propeller limits: Diameter: 91 inches, no reduction. Permitted Pitch setting at 30 degree, for 19.5 +/- 0.1 inches, no reduction. Feather: 85.0 +/- 0.5 inches.</td>
<td>07/03/2014</td>
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<tr>
<td>10048398</td>
<td>P3 ENGINEERING GMBH</td>
<td>BD100-1A10 CHALLENGER 300</td>
<td>Airfoil Sw/fBroadband Salcom installation and WFI Networked/cockpit. The change consists in the installation of a Salcom System with Sw/Broadband capability. The Salcom antenna is installed below the fin cover on the vertical stabilizer. In addition, a new cabin network is installed without additional modification.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements Special Condition (SC) F-03 “Security Protection of aircraft systems and network” issue 2 dated 06/30/2014. The requirements for environmental protection and the associated...</td>
<td>- Master Document List P5V-00024-MLD-001B dated 07/03/2014, Aircraft Flight Manual Supplement 23-13-09, reference P5V-00024-001B.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third p...</td>
<td>07/03/2014</td>
<td>Active</td>
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<tr>
<td>10048406</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td></td>
<td>550 BRAVO</td>
<td>P-RNAV Capability within GNS-XLS Single or Dual FMSP-RNAV approval of GNS-XLS FM8, including RNP10, RNP5, RNP1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Cessna Düsseldorf Airplane Flight Manual Supplement, 2013-18, 2013-18, Rev IR, dated 29 October 2013. Cessna Düsseldorf Installation Procedures, 2013-18-1, Rev IR, dated 29 October 2013, or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/03/2014</td>
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<td>EASA STC No.</td>
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<td>10048424</td>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td>FAA STC 03081AT</td>
<td>737-600-700X-800 737-300-400X-500</td>
<td>Installation of an emergency Locator Transmitter (ELT) system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Drawing List 2484- MDL Rev B dated 04/02/2007 Flight Manual Supplement 2484-AFS initial issue later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions off this decision) and/or the Technical Implementation Procedures (TIPs) applicable to this certificate/approval.</td>
<td>10/03/2014</td>
<td>Active</td>
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<td>10048426</td>
<td>ANORA INTERNATIONAL LLC</td>
<td>FAA STC ST04068A</td>
<td>B737-400</td>
<td>Installation of a main deck cargo loading system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Controlled Data List D18610 Rev C Dated Nov 05/11 Instruction for Continued Airworthiness report 6592 Rev A datedJune25th, 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or this ( \text{Rev.} ).</td>
<td>10/03/2014</td>
<td>Active</td>
<td>10048426</td>
</tr>
<tr>
<td>10048469</td>
<td>SPECTRUM AEROMED, LLC/ t/a SPECTRUM AEROMED</td>
<td>REV 1</td>
<td>FAA SAO3314CH</td>
<td>525, 525A, 525B, 525C</td>
<td>Air Ambulance Conversion/ Ambulance Conversion - Textron 525</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. According to Spectrum Aeroned Master Document List No. 6049DL, or later revisions, to be operated in accordance with AFM/Document No. 6049F/PM0. To be maintained i.e. ICA Document No. 6049F/PM0 later revisions of the above listed documents approved by EASA.</td>
<td>According limitation section of AFM/SP Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/04/2019</td>
</tr>
<tr>
<td>10048472</td>
<td>BRISTOW U.S. L.L.C.</td>
<td>FAA SRO9599RC</td>
<td>S-92A</td>
<td>Installation of a TDR-94D Transponder with ADS-B I/R</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Flight Manual Supplement RMS-SSA-2013-001 Rev I/Rev 1 for later revisions of the above listed documents approved by EASA Master Document List 5827X-TMP-MDL-0 Rev. EMaster Installation List 5827X-MDL-0 Rev. EMaster installation Manual 592-00-1000 Rev 300 Installation Manual 592-00-1000 Rev 300 Installation Manual 592-00-1000 Rev 300 Instruction for Continued Airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/04/2019</td>
<td>Active</td>
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<tr>
<td>STC Holder</td>
<td>Description</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
<td></td>
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<tr>
<td>DASSAULT AEROSPACE DESIGN FACILITIES LTD.</td>
<td>Installation of Spectrolab SX7 Searchlight as defined by ADF2013-MDL-100</td>
<td>ADF355 Fight Manual Supplement No. 12 &quot;SPECTROLAB SX7 STAR/SEARCHLIGHT INSTALLATION&quot;, Rev. (or later revisions of the above listed documents approved by EASA).</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product. The installation of this modification by third p...</td>
<td>13/02/2014</td>
<td>Active</td>
<td>10048467</td>
<td></td>
<td></td>
</tr>
<tr>
<td>ASI INNOVATION</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental Type Certificate Data List ref. ASI-SDL-0148-02 dated 13/03/2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Applicability limited to ERJ 190-100 LR MSN 130</td>
<td>14/03/2014</td>
<td>Active</td>
<td>10048483</td>
<td></td>
<td></td>
</tr>
<tr>
<td>TRACES AIRCRAFT CERTIFICATION ENGINEERING SERVICES</td>
<td>Installation of a 145/165 passengers seating arrangement.</td>
<td>Master Data List TS1115MDL Rev A dated January 07th, 2013/Instructions for Continued Airworthiness TS1115SR07 Rev I dated August 30th, 2013/later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2014/04/CF (or subsequent revisions of this decision) and/or the...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>14/03/2014</td>
<td>Active</td>
<td>10048504</td>
<td></td>
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<tr>
<td>DASSAULT FALCON SERVICE</td>
<td>Installation of LH Galaxy Bulb/Head installation and aircraft curtain modification.</td>
<td>DFS Change Approval Sheet dec. DFS7022-CAS Issue A dated 13March 2014/DFS Modification Sheet dec. DFS7022 Issue A dated 13 March 2014/later revisions of the above listed documents approved by EASA.</td>
<td>S/N 108 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>17/03/2014</td>
<td>Active</td>
<td>10048508</td>
<td></td>
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<tr>
<td>UMLAUT ENGINEERING GMBH</td>
<td>Activation of ADS-B Function using Mode S Extended Squitter</td>
<td>Umlaut Engineering GmbH Master Document List No. P3-4A002-MDL-001 Issue 6 dated 29/10/13 or later revisions of the above listed documents approved by EASA.</td>
<td>Refer to MDL for aircraft aftically; Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>17/04/2020</td>
<td>Active</td>
<td>10048510</td>
<td></td>
<td></td>
</tr>
<tr>
<td>BRISTOW HELICOPTERS (EASA.R.S.0192)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td></td>
<td>18/03/2014</td>
<td>Active</td>
<td>10048530</td>
<td></td>
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<tr>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC STO4038AT-D</td>
<td>G6-SP</td>
<td>Installation of an Airplane Compliation.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements Equivalent Safety Findings(1): CR- D-50 issue 1 Cabin Egress and (2) version 1. The requirements for environmental protection and the associated certificated noise or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>- FAA STC STO4038AT-D, issued 4th March 2014, FAA Statement of Compliance, ref. 1004-14-081, dated 20th March 2014, Gulfstream Aerospace Corporation Index List GDL10000006, Rev. - dated 4th March 2014, EASA Validated Configuration List, ref. GCD00000115, Rev. - dated Fifth March 2014, G6-SP Airl.</td>
<td>It is the installer responsibility to ensure that all STCs fitted and/or EASA approved to installation of this design change must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce roadworthiness effect upon...</td>
<td>18/03/2014</td>
<td>Active</td>
</tr>
<tr>
<td>IKHANA GROUP, LLC</td>
<td>REV.</td>
<td>FAA STC SAU04537LA</td>
<td>DHC-6 SERIES 100</td>
<td>Install and maintain DHC-6 Re-Life Flight Controls in accordance with FAA STC SAU04537LA.Reviews DHC-6 Re-Life Flight Controls are approved replacements for VikingAv Limited (formerly Bombardier Inc. or De Haviland Inc.) flightcontrols as listed in Annex 1 Rev. 1: Name Change from &quot;Ikhana Group Inc.&quot;</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements:The certification basis for parts changed or affected by the changes along with the reference date of the applicable, November 9, 2007, is based upon FAR Part 23 as amended by Am...</td>
<td>Master Data List, Document No. ER21-1076 Revision A, dated April 29, 2013 Drawing List, Document No. ER 218-1077 Revision B, dated March 25, 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA EC Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>18/06/2021</td>
<td>Active</td>
</tr>
<tr>
<td>FEDERATION FRANCAISE DE VOL A VOILE</td>
<td>1</td>
<td></td>
<td></td>
<td>Matraillon sur planeur Pegase</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>Supplement Manuel de Maintenance Pegase CAB61/2014Supplement Manuel de vol Pegase CAB48/2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change and any other previously installed design change and/or repair will introduce roadworthiness effect upon the original product.</td>
<td>18/03/2014</td>
</tr>
<tr>
<td>LUFTHANSA TECHIK AG</td>
<td>A320-212 / -213 / -231 / -232 / A321-111 / -112 / -131 / -211 / A320-211 / -212 / A320-214 / -215 / -216 / A320-211 / -111 / -112 / -131 / -133 / A319-115 / -131 / -132 / -133</td>
<td>ACARS CMU software upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>- Declaration of Compliance ADC 3530, dated 13/03/2014, Supplemental Type Definition Document STD023-1400M, Issue 01 Rev.00, dated 13/03/2014, Compliance Record List CRL23-1400M, Issue 01 Rev.0, dated 13/03/2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce roadworthiness effect upon the original product.</td>
<td>18/03/2014</td>
<td>Active</td>
<td>10048532</td>
</tr>
<tr>
<td>AIRBUS AIR ESPACE</td>
<td>500, 550, 560</td>
<td>Change of Weather Radar and indicator on Cessna 500, 550 and 560.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>Airlec Air Espace, Instructions for Continued Airworthiness, AEE 06SICA, rev. 1, dated 18 March 2014 Airlec Air Espace, STC Data List, AEE 056SIL, rev. 2, dated 18 March 2014, later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce roadworthiness effect upon the original product.</td>
<td>20/03/2014</td>
<td>Active</td>
<td>10048548</td>
</tr>
<tr>
<td>SAA SOLUTIONS FOR AVIATION, S.L</td>
<td>CL-600-2B16</td>
<td>Medical berth installation in CL-604</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements:CS25 Amnd. 1225-D025-603 a) (b) (25.603 a) 25.607 a) 25.609 a) 25.610 a) 25.1309 a) 25.1357 a) b) (c) The requirements for environmental protection</td>
<td>SAA Master Drawing List No. 80404 RPT10 Issue 1 dated 9th May 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>Restricted to ADSN 5306, 5397 and 5416 Prior installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce roadworthiness effect upon the original product.</td>
<td>20/03/2014</td>
<td>Active</td>
<td>10048551</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Rev.</td>
<td>Foreign/NA A STC no.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10048557</td>
<td>AIRGLAS, Inc.</td>
<td>FAA STC SH1910NM</td>
<td>412, 412CF, 412EP</td>
<td>Fabrication and installation of Airglas, Inc. PN L6500A-1 or L6500R-raft ski and PN L6500A-2 or L6500R-raft ski in accordance with Airglas, Inc. Master Drawing List No. L6500A/R.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EASA Approved Flight Manual Supplement for Bell Model 412J/Model 412EP, and Model 412F helicopters, L6500AR of the revisionlevel specified in the attached table or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004042CF (or subsequent...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise/radiation effect upon the airworthiness of the product.</td>
<td>21/03/2014</td>
</tr>
<tr>
<td>10048558</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC ST0200SE</td>
<td>A330-200</td>
<td>Interior Reconfiguration of Qatar Airways A330-200 installation of SPG Business Class Mini-Pod suites, Recaro tourist classseats, closets, class divider, storage bins and installation list. Referto Certification Plan B2FS225-D10-EASA, rev A for additional information. Note 1: ISGPS and IFE / GCS and WLAN systems.</td>
<td>The Certification Basis for the original product as amendedby thefollowing additional or alternative airworthiness requirements Refer to The Compliance Checklist in Certificationplan B2FS225-D10-EASA rev A, which presents the list of paragraphs at later CS Amendment level. The requirements for en...</td>
<td>- Certification Plan B2FS225-D10-EASA, rev A. Master Drawing List, ref B2FS225-D01, rev N ICA - Supplement To Instructions For Continued Airworthiness FIRDocument, ref B2FS225-D15, rev IR (Note: MMEL is out of scope for theEASA approval)+ FAA Letter 1505-14-72, dated on 20th March 2014 later re...</td>
<td>1) Prior to installation of this modification it must be determined that the interrelationship between this modification and/or repair will introduce no adverse effect upon the airworthiness of the product(2) ICA must be incorporated into the operator’s ma...</td>
<td>21/03/2014</td>
</tr>
<tr>
<td>10048564</td>
<td>AVIATION COMPLETION INNOVATIONS INC</td>
<td>FAA STC SR030606N</td>
<td>AB139, AW139</td>
<td>Installation of Engine Wash Kit in either Configuration 1 orConfiguration 2</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SR030606N. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the or...</td>
<td>Installation Drawing Lists - Configuration 1: Air Concepts International, Inc. Report No.41252-405-99, Revision IR, dated 7 November 2011 - Configuration 2: Aviation Completion Innovations, Inc. Report No.IL00139.12-2, Revision IR, dated 23 August 2018.Instructions for Continued Airworthiness - When...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2019</td>
</tr>
<tr>
<td>10048570</td>
<td>ASI INNOVATION</td>
<td>FAA STC</td>
<td>E90</td>
<td>Radome with relay antenna installation and provision for electrical system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>ASI-SDL-0184-02 rev.3 dated 21.03.2014To be operated in accordance with ASI-AM-0184-01 rev. 1 dated 21.03.2014No change to the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>As per the Flight Manual Supplement, &quot;Other&quot;: limited to MSHLW 32B and LW 278. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/03/2014</td>
</tr>
<tr>
<td>10048589</td>
<td>GAMA ENGINEERING Ltd</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B300</td>
<td>Installation of Engine Exhaust Cowling</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Gama Engineering Modification HL M01162 and Installation procedures ref IP-1162 Issue 2.Operation in accordance with Gama Engineering Aircraft Flight Manual Supplement AFMS-1162-01 Issue 1.Instructions for Continued Airworthiness defined in Gama EngineeringReport ICA...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/03/2014</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>JET AVIATION AG</td>
<td>REV. 3</td>
<td>737-800ER</td>
<td></td>
<td>VIP Interor Installation B737-800ERs defined in the –Project Description, Doc. No.: BBJ37546-001. – Master Data List, Doc. No.: BRJ37546-001 Revision 1: TSO-related Imitation removed.Revision 2: transfer of certicate from “JET Aviation AG, Basel” (DOA21,356) to “JET Aviation AG, Basel”.</td>
<td>The Certification Basis for the original product as amended by alterative airworthiness requirements: - Special Condition (s): CRI-D-02 Installated Compartments.CRI-D-10 Glass Screens of Displays / Monitors CRI-D-14 Shower Installation – Equivalent Safety Finding(s): CRI-D-09 I...</td>
<td>- Master Data List, Doc. No.: BBJ37546-001 Rev IR- Aerospace Fight Manual Supplement, Doc. No.: BBJ37546-201-01 Rev A- Instructions for Continued Airworthiness, Doc. No.: BBJ37546-150-01 Rev A- EWIS Inructions for Continued Airworthiness, Doc. No.: BBJ37546-150-02 Rev I()r latest revisions of the a...</td>
<td>01/02/2017</td>
<td>Active</td>
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<tr>
<td>J &amp; C ENGINEERING SERVICES Ltd.</td>
<td>REV. 3</td>
<td>CA UK RESPONSIBIL ITY</td>
<td>737-300</td>
<td>Deactivation and removal of cargo loading system on QC aircraft withreconfiguration to a standard LOPA.</td>
<td>The Certification Basis for the original product as amended by alterative airworthiness requirements: - Special Condition (s): CRI-D-02 Installated Compartments.CRI-D-14 Shower Installation – Equivalent Safety Finding(s): CRI-D-09 I...</td>
<td>J ()&lt;-C Engineering Services Ltd Modification No.: JCMDM3X746 Issue 2 dated 11 March 2014 for latest revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined thalte relationship between this design change and any otherpreviously installed design change and or repair will introduce noadverse effect upon the airworthiness ofthe product. The installation of this modification by third p...</td>
<td>25/03/2014</td>
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<tr>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 1</td>
<td>FALCON 900EX (F900EX)</td>
<td></td>
<td>Seat Change on Dassault Falcon 900EX (SN998).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged</td>
<td>Master Document List ref. MDL016Z001 issue A, dated 26/3/2014()r later revisions of the above listed documents approved by EASA.</td>
<td>Applicability limited to Falcon 900EX SN998 Prior to installation of this design change it must be determined that the deviation between this design change and any otherpreviously installed design change and or repair will introduce noadverse effect upon the airworthiness ofthe product.</td>
<td>27/09/2018</td>
</tr>
<tr>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 1</td>
<td>FALCON 900EX (F900EX)</td>
<td></td>
<td>Cabin Refurbishment with replacement of CMS /IFE and CabinLighting onDassault Falcon 900EX (SN998).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged</td>
<td>Master Document List ref. MDL017Z001 issue A, dated 21/03/2014()r later revisions of the above listed documents approved by EASA.</td>
<td>Applicability limited to Falcon 900EX SN998 Prior to installation of this design change it must be determined that the deviation between this design change and any otherpreviously installed design change and or repair will introduce noadverse effect upon the airworthiness ofthe product.</td>
<td>27/09/2018</td>
</tr>
<tr>
<td>airBus engineering GmbH</td>
<td>REV. 2</td>
<td>695, 695A, 695B, 690, 690A, 690B, 690C, 690D</td>
<td></td>
<td>Installation of single or dual Garmin G600 Avionics System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged</td>
<td>Flight Manual Supplement PO-1569-00000, Issue 1-6- Description of Modification DE-1569-00000, Issue 1, or later approved revisions- Maintenance Manual Supplement MM-1569-00000, Issue 1, or later approved revisionsor later...</td>
<td>The aircraft must have been already equipped with a suitable GPSreceiver.Prior to installation of this design change it must be determined that the deviation between this design change and any otherpreviously installed design change and or repair will introduce noadverse effect upon the airworthiness of the airframe...</td>
<td>10/02/2021</td>
</tr>
<tr>
<td>DIAMOND UPGRADES, L.L.C.</td>
<td></td>
<td>BELL 212</td>
<td></td>
<td>Installation of Pratt &amp; Whitney PT6T-3DE or PT6T-3DF Twin-Pac to replace PT6T-3E or PT6T-3F Twin-Pac(While the FAA approved revision is granted for both the -3DE and -3DF, the EASA/DOA of investigation and approval is limited to the -3DF as the 3DEengine is currently not EASA approved.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged</td>
<td>RFM Supplement, AU-001206, Rev. 5, dated 31/07/2013or laterrevisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/44CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EUUISA Bilateral Agreement installation Inst...</td>
<td>This approval is limited to the installation of PT6T-3DF Twin-Pac only.PT6T-3DF engine is currently not EASA certified. The Bell TechinicallBulletin listed in the installation Instructions AU-001207, Rev. 4 dated 26/7/2013, or later FAA approved revisions, must be cross-referenced accordingly. Prior to...</td>
<td>26/03/2014</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10048634</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>A380-841/-842/-861</td>
<td>Installation of non-electrical Floor Path Marking System.Rev. 1: Extends applicable pre-change configurations to include non-electrical or electrical floor path marking systems</td>
<td>The Certification Basis for the original product as amended to include the following additional or alternative airworthiness requirements:CS25 Amtt. 13 for the affected areas as defined in Ref. CRL33-16/355Issue 1. Revision 2:The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- OP M-2084-0029300-501 Master Flight Manual Supplement, Rev. R dated 19/02/2024.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect on the airworthiness of the product.</td>
<td>04/02/2017</td>
</tr>
<tr>
<td>10048638</td>
<td>AEROLITE AG</td>
<td>REV. 7</td>
<td>429</td>
<td>EMS Equipment P/N 429-0029300-501. History of revisions:Original Revision consists of modular cabin EMS configurations in which crew seats and up to two stretchers can be installed. A medical supply panel provides EMS oxygen and electrical supply. Stowage and attachment provisions...</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>For EMS Equipment P/N 429-0029300-501 Master Document List, MDL-13-008 Rev. P, Flight Manual Supplement, FMS-429-0029300-501 Rev. P, dated APR 2020.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect on the airworthiness of the product.</td>
<td>19/02/2024</td>
</tr>
<tr>
<td>10048639</td>
<td>KUERZI AVIONICS AG</td>
<td>REV. 1</td>
<td>M20K</td>
<td>Certification of RNAV approach down to LPV minima.Compliance to AMC20-28 that the installed equipment is suitable for RNAV (GNS) Approaches down to LPV Minima. Compliance to AMC20-27, resp. CRI F-10 (Airworthiness Criteria for RNAV Equipment supporting APF BARO VNAV approaches to LNAV/VNAV)...</td>
<td>FAR 23 Amtt. 6. The Certification Basis for the original product as amended to include the following additional or alternative airworthiness requirements:For this non significant change, compliance to FAR 23, Amtt. 6 was shown, in particular 23.1301, 23.1325, 23.1329, 23.1339 and 23.1417. The certification remains unchanged and remains applicable to this certificate/approval.</td>
<td>Master Document List, MDL-201313468, Rev. NC, dated 26.03.2014.</td>
<td>Required Pre-modifications - As a minimum dual Garmin GNS430W and a Garmin G500 Data Link Display must be installed and approved. An independent 3rd party should be included in the installation of this design change.</td>
<td>01/04/2014</td>
</tr>
<tr>
<td>10048649</td>
<td>365 AEROSPACE DQS LIMITED</td>
<td>REV. 4</td>
<td>737-8i-8200-9, 737-600-700-800-900-900i-900i</td>
<td>Introduction of Controller-Display Data Link Communication.Rev. 2: Transfer of STC from 365 Aerospace Ltd. Rev. 3: Models 737-8, 737-8200, 737-9 added to the applicability list Rev. 4: Correction the associated technical documentation.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s) CRI F-0 - Data Link Services for the Single European Sky Issue 2014 - Flight Recorders including Data Link Recording Issue 3.</td>
<td>For models 737-600-700-800-900-900i-900i: - Master Data List and Instructions for Continued Airworthiness - MOD-1446 Rev. 4A, dated 01/04/2014.</td>
<td>Required Pre-modifications - As a minimum dual Garmin GNS430W and a Garmin G500 Data Link Display must be installed and approved. An independent 3rd party should be included in the installation of this design change.</td>
<td>01/04/2014</td>
</tr>
<tr>
<td>10048650</td>
<td>CESSNA DÜSSELDORF CITATIONSERVIC E CENTER gmbh</td>
<td>CITATION 550 B BRAVO</td>
<td>P-RNAV approval of Universal UNIS-1Cap or UNIS-1K</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Cessna Düsseldorf Master Document List, 2013-6-16, rev. R, dated 10/March 2014 - Cessna Düsseldorf Aircraft Flight Manual, 2013-8-6, rev. A, dated 31/October 2013 - Cessna Düsseldorf Installation Procedure, 2013-8-1, rev. R, dated 28/June 2013 or later revisions of the above listed documents approved by EASA...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect on the airworthiness of the product.</td>
<td>27/03/2014</td>
<td>Active</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>10048661</td>
<td>SERAM - SOCIETE D'ENTRETIEN EPREPARATION AERONAUTIQUE MACONNAISE</td>
<td>T891, T810, TB20</td>
<td>Dual installation of Garmin GNS 400/500 series system and associated aircraft electronics, and of Garmin GMA 340 audio system, according to SERAM modification No. 412.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and / or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>SERAM Flight Manual Supplement n° 412-3 - TB20/1020 - Use of GNS4xx/Avionics AW AV in IFR B/RNAV, P/RNAV, RNP approach LN Nav and LPV, revision 0 dated 30/10/2013. SERAM Flight Manual Supplement n° 412-1 - TB9/1020 - Use of GNS4xx/Avionics VFR, revision 0 dated 30/10/2013.</td>
<td>Initial installation must be performed by authorised SERAM personnel. This installation is approved for VFR and for IFR B/RNAV, P/RNAV and RNP approach LN Nav operations. It is also approved for RNP approach LPV operations but only with GNS430 WA/AV or GNS 530 WA/AV systems enabling GNSS vertical guidance.</td>
<td>28/03/2014</td>
<td>Active</td>
</tr>
<tr>
<td>10048669</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A380-841/842-861</td>
<td>Installation of Data Broadband Transceiver System with full provision for On-Board Mobile Telephony System</td>
<td>The Certification Basis for the original product as amended by additional or alternative airworthiness requirements or Equivalent Safety Findings(s). CRI C-02 - Vibration Buffeting and Flutter Compliance Criteria. The requirements for environmental protection and the associated certified noise.</td>
<td>Supplemental Type Definition Document: STD044-1301M Iss. 01Rev.01Rev.01Instructions for continued Airworthiness: A-00-1351/ICA Iss.01 Rev. 00XML-Supplement: A-00-1351-AFM Iss.02 Rev.00Compliance Record List: CRL44-1301M Iss.01 Rev.09or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>31/03/2014</td>
<td>Active</td>
</tr>
<tr>
<td>10048670</td>
<td>AEROTECH SARL</td>
<td>TB 20</td>
<td>Installation of two Garmin GTN650 systems, a kind KIN32 OME and a Garmin GMA340 audio system, according to Aerotec modification no. MOD 13528</td>
<td>The Certification Basis for the original product as amended by additional or alternative airworthiness requirements CSV 23 Arrt. 3 § 867, 1301, 1309, 1311, 1321, 1322, 1327, 1329, 1351, 1353, 1357, 1365, 1381, 1431, 1529, 1614, 1947, 1581The requirements for environmental protection and...</td>
<td>Aircraft Flight Manual Supplement No. AE13258SMD revision 0 dated 12/13/2013</td>
<td>This installation is limited to TB 20 S/N 1245. This installation is approved for IFR B/RNAV, P/RNAV, RNP approach LNAV and LPV operations, with coupling to KAP 150 autopilot and Auen 1000P display system. Introduction of new functions through software upgrade must be subject of another EASA approval.</td>
<td>31/03/2014</td>
<td>Active</td>
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<tr>
<td>10048697</td>
<td>INFILIGHT WARNING SYSTEMS, Inc.</td>
<td>FAA ST0258AAT</td>
<td>767-200, 767-300</td>
<td>Installation of the inflight warning Systems Inc., Fan Vibration Monitor Unit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and / or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Drawing List NDL 211202 Rev A dated May 6, 2002Master Data List 13415 Rev BInstructions for Continued Airworthiness ICA 13105 Rev R dated Sept 9, 2013or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04CF (or subsequent revisions ofth...</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and / or repair will introduce a degrading effect upon the airworthiness of the product.</td>
<td>31/03/2014</td>
</tr>
<tr>
<td>10048698</td>
<td>INFILIGHT WARNING SYSTEMS, Inc.</td>
<td>FAA ST0253AAT</td>
<td>767-300F, 767-400ER, 767-200, 767-300</td>
<td>Installation of an air conditioning fan monitors to monitor fan bearings performance.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and / or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Drawing List 213300 Rev D dated January 13, 2004Master Data List 13415 Rev B dated December 21, 2005Instructions for continued Airworthiness ICA 13149or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04CF (or subsequent revisions ofth...</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and / or repair will introduce a degrading effect upon the airworthiness of the product.</td>
<td>31/03/2014</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Rev. Level</td>
<td>Foreign/NAA Code/Ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>BOMBAIRDIER INC.</td>
<td>10048730</td>
<td></td>
<td>TC-CA-C-1602-02-20D</td>
<td>BD-700-1A10</td>
<td>Aircraft Completion: 14 passenger interior configuration - EASA validation of Transport Canada STC C-LSA02-202D at issue 1</td>
<td>The Certification Basis for the original product as amended by the following: Additional or alternative airworthiness requirements. Deviation(s): EASA Deviation CRI-D-100 at issue 3 &quot;Doors between passenger compartments&quot;. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Modification No. 3493-017, dated 26th March 2014 and EASA - The STC is limited to aircraft serial number 9086. Refer to the required Flight Manual Supplements and Airworthiness limitations. STC supplements listed in associated technical documentation - Bombardier Service Bulletin 700-31-018 (Modification - Integrated Avionics Systems (IAE) – Avionics 2001)</td>
<td>31/03/2014</td>
</tr>
<tr>
<td>AEROSPACE DESIGN AND ENGINEERING INSTITUTANTS LTD</td>
<td>10048705</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>A320-214</td>
<td>Removal of Aircraft Cabin Interior and Associated Changes.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aerodata Modification No. 1401, dated 14 April 2014 and later reissues of the above listed documents approved by EASA.</td>
<td>Restricted to: ARNs 4855, 4851, 4946. The following Aeroscan Modifications are required to be concurrently onboard as part of this STC: MOD-1402 - Removal of Satellite TV System Provision No 1403 - Removal of Gogo Aireal Broadband Service No 1404 - Removal of Windows &amp; Class Dividers MOD-1405 - Re...</td>
</tr>
<tr>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>10048730</td>
<td></td>
<td>DA42</td>
<td>DA42</td>
<td>Transfer of STC 10048730 from Crosby Aviation (Jersey) Ltd to Cranfield Aerospace (Jersey) Ltd/CRS.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Printed document, DT- CAeM/019-01, dated 12.01.2015 and later revisions of the above listed document(s) approved by EASA.</td>
<td>Prior to installation of this change in order to maintain the interrelationship between this change and any other previously installed change and/or repair and any interrelationship between this change and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>AERODATA AG</td>
<td>10048730</td>
<td></td>
<td>B300</td>
<td>B300</td>
<td>Installation of AD-AFIS-112 into B300 with ProLine 2 Cockpit. The...</td>
<td>The Certification Basis for the original product remains applicable to items not affected by the modification, for items affected by the modification follow the new Certification Basis is elected to comply with: CS-23 AMT 3, except 23.561 and 23.786 (Rev 0) for Continuation Airworthiness Aircraft Maintenance Manual Supplement, Aerodata document AMM-141 Rev 0, for each Aircraft Maintenance Manual Supplement, Aerodata document AMM-141 Rev 0, for each</td>
<td>Installation of AD-AFIS-112 in accordance with Master Document List, Aerodata document MDA-355 Rev 2, dated March 20, 2014.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>CRANFIELD AEROSPACE SOLUTIONS LTD</td>
<td>10048736</td>
<td></td>
<td>A340-311-312-313</td>
<td>A340-311-312-313</td>
<td>Installation of the Rack, Cable and Server for Tracking and Data Communications.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Cranfield Aerospace Modification No. CAW/340/13740/1/2014Aerospace Ltd Type Design Definition Amendment No. CADD/DR1177 Issue B dated 29 February 2014 and later reissues of the above listed documents approved by EASA.</td>
<td>Installation of EASA STC 10048736 is a prerequisite for this modification.</td>
</tr>
<tr>
<td>AVIONITEC AG</td>
<td>10048745</td>
<td></td>
<td>PA-31T, PA-31T1, PA-31T2, PA-31T3</td>
<td>PA-31T, PA-31T1, PA-31T2, PA-31T3</td>
<td>Dual GNS 530W upgrade incl. LPV Certification Removed - Dual Garmin GNS 530 (GPS/COM/AHRS) Installed - Dual Garmin GNS 530W (GPS/COM/NAV) Compliance showing to AMCDO-27, resp. CRI F-100...</td>
<td>FAR 23 AMT 7: The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List: No. 3493-017, dated 11.01.2014 and later instructions for continued airworthiness: No. 3493-016, dated 19.11.2013 and for the above modifications Instruction: No. 3493-050, dated 11.01.2014 and later reissues of the above listed documents approved by EASA.</td>
<td>The approval holder shall fulfill the obligations of Part 21 §1.11.03(f) for installation of this modification: the STC holder shall fulfill the obligations of Part 21 §21.113(b) for installation of this modification: the STC holder shall fulfill the obligations of Part 21 §21.113(b) for installation of this modification: the STC holder shall fulfill the obligations of Part 21 §21.113(b) for installation of this modification: the STC holder shall fulfill the obligations of Part 21 §21.113(b) for installation of this modification: the STC holder shall fulfill the obligations of Part 21 §21.113(b) for installation of this modification: the STC holder shall fulfill the obligations of Part 21 §21.113(b) for installation of this modification:</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10048773</td>
<td>RUAG AG</td>
<td>REV. 2</td>
<td></td>
<td>PC-12, PC-12/45, PC-12/47</td>
<td>PC-12 Antenna Installation. The initial STC approved multiple antenna installations Rev. 1 of this STC provides multiple antenna installations and locations options.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Instructions for Continuous Airworthiness Doc. No. ICA-800156 Rev. D; Master Data List Doc. No. MDL-8001143 Rev. C; other revisions of the above listed documents approved/accepted under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<td>10048774</td>
<td>SCHEDEL ENGINEERING Ltd</td>
<td></td>
<td></td>
<td>SH Tail Aero Modification</td>
<td></td>
<td></td>
<td>Technical Note TN 4. Rev 1 SH Tail Aero Modification Certification Document CD, VT Rev 1 for later revisions of the above listed documents approved by EASA.</td>
<td>Valid only for Types listed in the Technical Note ScheDEL Engineering TN 4Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<td>10048781</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd</td>
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<td>CAA UK RESPONSIBILITY</td>
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<tr>
<td>10048788</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td></td>
<td>767-400ER</td>
<td>&quot;Stirred&quot; VIP cabin completion-Variant 1: baseline cabin design. Variant 2: baseline cabin design modified with removal of one interior door, redesignation of one flight attendant seat for in-flight use only, replacement of certain glass mirrors with plastic sheeting.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List Report No: BAC-864ML02 Issue 2, dated 02 April 2014 Instructions for Continued Airworthiness Report No: BAC-864CAI2 Issue 1, dated 11 November 2013</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10048789</td>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>REV. 2</td>
<td></td>
<td>B767-200-300</td>
<td>Activation of GSMConnex system installation</td>
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<td>10048790</td>
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<td>EASA STC No.</td>
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<td>Rev. Level</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10048793</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SAB8996AC-D</td>
<td>F35, E33C, E33C, 35-33A, 35-33B, 35-33C, 35-33E</td>
<td>Installation of S-TEC System 55/55X Two Axis Automated Flight Guidance System (Model ST-55X)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- FAA-CT-003415-A issued January 14, 2014.</td>
<td>- FAA STC ST03974AT is not part of this approval as STC ST03974AT is not being concurrently validated. Prior to installation of this design change it must be...</td>
<td>10/04/2014</td>
</tr>
<tr>
<td>10048796</td>
<td>VHI AEROSPACE Inc.</td>
<td>REV 1</td>
<td>TCCA STC SH09-43</td>
<td>EC135 T2+ EC135 T3 EC135 T2+ EC135 SH09-43</td>
<td>Installation of Crew Bubble Window in accordance with TCCA STC SH09-43.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval.</td>
<td>- FAA-CT-003415-A issued January 14, 2014.</td>
<td>26/01/2022</td>
</tr>
<tr>
<td>10048797</td>
<td>CENTEX AEROSPACE, INC.</td>
<td>REV 2</td>
<td>FAA STC SA11010SC</td>
<td>2.B200GT, 2.B200, B200C, B200CT</td>
<td>Halo 250 IGW is not applicable to this STC.</td>
<td>Halo 250 IGW is not applicable to this STC.</td>
<td>- Referral to Centex Aerospace Drawing No. 006-0000-1000, Rev. M, or, for the approved weight limits and that it be part of the STC.</td>
<td>17/10/2018</td>
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<td>STC Holder</td>
<td>STC Ref.</td>
<td>Rev.</td>
<td>Foreign/NAA STC Ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>ATELIER CHABORD</td>
<td>10048838</td>
<td>REV 3</td>
<td></td>
<td>DR 400/160, DR 400/160D</td>
<td>Installation of exhaust system according to Chabord modification ST/C/DRA02-2012-01 REV 3. Enhanced silent blocks. Alternative anti-corrosion surface treatment.</td>
<td>The Certification Basis for the original product as amended by the following: Additional or alternative anti-airworthiness requirements.</td>
<td>Technical Design Document ST/C/DRA02-2012-01 REV 3.</td>
<td>03/03/2022</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>10048807</td>
<td></td>
<td></td>
<td>DR 400/140, DR 400/140B</td>
<td>Conversion of cabin layout from 144 PAX to 134 PAX. Dual class.</td>
<td>The Certification Basis for the original product as amended by this certificate/approval.</td>
<td></td>
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<td>MID-CONTINENT INSTRUMENT Co, Inc</td>
<td>10048833</td>
<td>REV 2</td>
<td>FAA SAD199998E</td>
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<td>MT-PROPeller ENTWICKLUNG GmbH</td>
<td>10048838</td>
<td>76</td>
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<td>MTY-12-B-C-FCF(1)-1163-506 ow/proch 76</td>
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<td>STC Holder</td>
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<td>Foreign/NAA Code(s)</td>
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<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
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<td>IKHANA GROUP, LLC</td>
<td>10048850</td>
<td>REV. 1</td>
<td>FAA STC SA02474AL</td>
<td>DHC-6 SERIES 100, 200, 300</td>
<td>Install and Maintain DHC-6 Rolls of engine nacelles by replacement of fatigue- susceptible structural components.RWMI DHC-6 Rev. Nacelles are approved replacements for V1450 Air Canada (Air渤海) or De HavillandRev nacelles as listed below.</td>
<td>The Certification Basis for the original product as amended by this modification is that it must be determined that the design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>18/06/2021</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>10048858</td>
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<td>Installation of external cameras</td>
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<td>TELAIR INTERNATIONAL Inc.</td>
<td>10048864</td>
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<td>FAA STC ST02489LA</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Light Weight LD3 Cargo Container</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
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<td>SUPERJET INTERNATIONAL S.P.A.</td>
<td>10048865</td>
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<td>C27-J</td>
<td>Dual EFB (class 2) installation on C27J</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
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<td>LIVETV, L.L.C.</td>
<td>10048878</td>
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<td>FAA STC ST02351CH</td>
<td>Installation of the LiveTV Ka-Band Connectivity System</td>
<td>The Certification Basis for the original product as amended by this modification is that it must be determined that the design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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**List of EASA Supplemental Type Certificates**

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<th>EASA STC No.</th>
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<th>Rev. Level</th>
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<tr>
<td>10048882</td>
<td>AVIATIONTEC AG</td>
<td></td>
<td></td>
<td>P210N</td>
<td>Replacement of KX-155 VHF-COM/NAV No. 2 by a 2. GNS 430W Out. King XX-155 system - Garmin GNS-430W system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The certified noise and/or emissions levels of the original product were unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List No. 3150-017 Rev. of 21.02.2014 for Continued Airworthiness 3150-015 or 21.02.2014AMM Supplement No. 2259-011 Rev. 1 of 21.02.2014 for the above listed documents approved by EASA.</td>
<td>SN#: P210-0134. The approval holder shall fulfill the obligations of Part 21, 21.02.2014 for Continued Airworthiness 3150-015 or 21.02.2014AMM Supplement No. 2259-011 Rev. 1 of 21.02.2014 for the above listed documents approved by EASA.</td>
<td>22/04/2014</td>
<td>Active</td>
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<td>10048912</td>
<td>AVIATION FABRICATORS, Inc.</td>
<td>REV 1</td>
<td>FAA SA02488LA</td>
<td>B300, B300C, B200C, B200CT, B200T, B300, B200C, B200CT</td>
<td>Installation of Aviation Fabricators Aft Tuck Cabinet Seat (including oxygen, air and light systems).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA02488LA. The requirements for environmental protection and associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>1. FAA Approved Aviation Fabricators Master Data List, No. AF-483DML Rev. S dated 11 June 2015 or FAA STC 10046426c. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/11/2018</td>
<td>Active</td>
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<tr>
<td>10048922</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV 2</td>
<td>FAA STC ST03184NY</td>
<td>A330-200, -300</td>
<td>Installation of a Panasonic a32 In-Flight Entertainment System and Global Communication Suite System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirement: the following paragraph(s) at a later amendment Refer to section 3 of Compliance Summary Report Project results: 0664-0888-2356 Rev. 4, 4 of 2009 and 0664-0888-2356 Rev. 16 of 2016.</td>
<td>Delta Engineering Instructions for Continued Airworthiness No. 0754-0852B-2356 Revis 0664-0888-2356 Rev. 4/2016 Aircraft EASA STC 10046426c. The requirements for environmental protection and associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>This STC will require the following separate approvals as pre-requisites. a. T-PED tolerance approved by a separate STC: EASA STC 10045386b. The Panasoni QCS antenna (k-band) is installed by a separate EASASTC 10037167c. For configuration TL10741-01 Zodiac Aerospace Service Bulletin 0344L-25-11 and EASA STC 10046426c. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/07/2018</td>
<td>Active</td>
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<td>10048924</td>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SA09542AC-D</td>
<td>SERIALS FE-1 AND SUBSEQUENT</td>
<td>Installation of S-TEC 01304 Digital Flight Control System (DFCS).</td>
<td>As defined on FAA STC SA09542AC-D. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Delta Engineering Instructions for Continued Airworthiness No. 0754-0852B-2356 Revis 0664-0888-2356 Rev. 4/2016 Aircraft EASA STC 10046426c. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third p...</td>
<td>23/04/2014</td>
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<td>10048926</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>REV. 1</td>
<td>BD-100-1A10</td>
<td>ASPIRE 200 SATCOM SYSTEM INSTALLATION IN CHALLENGER 300</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>CL-301-SBB/1 Service Bulletin EASCL301-HDK01 Flight Manual Supplement EDCir later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/04/2014</td>
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<td>10048942</td>
<td>STC TWENTY ONE LIMITED</td>
<td></td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>F900EX</td>
<td>Introduction of Honeywell Ovation Select Cabin Management System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>STC Twenty One Master Document List No. S21.MXL-1170. Issue2 dated 02/04/2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third p...</td>
<td>25/04/2014</td>
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<td>STC Holder</td>
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<td>HELIONE (NORWAY) AS</td>
<td>MOD AHSE-44010,RNAV-2 FMS Installation.</td>
<td>- Heli-One Flight Manual Supplement N1HY-156 &quot;RNAV-2 INSTALLATION&quot; HSJ-6461 10&quot; Initial issue, dated 13 Feb 2014, or later revisions of the above listed documents approved by EASA. - Heli-One Master Drawing AHSE-44011 Rev. - Heli-One Embodiment Instructions AHSE-44017 Rev D. - Heli-One Certification.</td>
<td>Prior to installation of this modification it must be determined that the installation of this modification and any other previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/04/2014</td>
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<td>BOMBAKDIER INC.</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 9083</td>
<td>Transport Canada STC No. C-LSA62-2693D Issue 4 dated 12 December 2003.Aero Consulting Services Modification Data Summaries D3155 Rev B and D31300 Revision N/C. The following Bombardier Aerospace Flight Manual Supplements C- FSJ240003X Rev N/C Cabin Master Power Switches - FM2500048 Rev A Interior Complex.</td>
<td>Restricted to ASN 9083. Bombardier Service Bulletin 700-31-01Modification - Integrated Avionics Computers - Avionics 2001 Batch 1 3AC. Upgrade must be embodied. Prior to installation of this design change it must be determined that the installation the interrelationship between this design change and any other previous...</td>
<td>30/04/2014</td>
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<td>10049065</td>
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<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>Installation of seven (7) Upgraded Lavatories</td>
<td>Master Data Lists MDL 08T052L001(1)(4)(5)(6), revision L, dated September 14, 2012; and 10T078L001, revision IR, dated September 14, 2012; and 08T052L089, Revision A dated June 8, 2009; and 10T078R009, Revision B dated July 6, 2012 or later revisions of the above listed documents.</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third p...</td>
<td>30/04/2014</td>
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<td>RUAG AG</td>
<td>Omni Antenna &amp; Actuator Down Link System installation PIN 800-1693-02Rev 1 Transfer of STC to a new legal entity that is holder of a new DOA(EASA 21J 7601)</td>
<td>Rotorcraft Flight Manual Supplement No. FMS-60034 Revision AMaster Document List No. MDL-400168. Revision Allocation for Continued Airworthiness No. ICA-8001687. Revision Aor later revisions of the above listed document(s) approved/accepte...</td>
<td>S/N: 1045 &amp; 1098This installation only covers the mechanical Antibes provision/installation (including the electrical actuator); the installation of the equipment is not part of the electrical communication system. Prior to installation of this design change it must be determined that the installation of this modification by third party...</td>
<td>18/01/2021</td>
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<td>L2 CONSULTING SERVICES Inc.</td>
<td>Rev 0: Installation of an Iridium Communication System in accordance with Master Drawing List 2322-843-ML Rev ADRev 1: Addition of the PTA 12-100 Dailier in accordance with the Master Drawing List 2322-843-ML rev A</td>
<td>- Airplane Flight Manual Supplement, Ref. 2322-843-4AFM, Revision G dated 25 Sept, 2013 Instructions for Continued Airworthiness, Ref. 2322-843-ICA, Revision X, dated 28/04/2014 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequ...</td>
<td>Prior to installation of this design change it must be determined that the installation of this modification and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/02/2016</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>JCB AÉRO SA</td>
<td>STC Holder</td>
<td>A320-214</td>
<td>A320 cabin reconfiguration from 162 to 180 pax</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Project Data List A320-03444-01-PDL-01, Rev. A, dated 29/04/2014 - Compliance Report A320-03444-01-CR-01, Rev. IR, dated 26/04/2014 - Instructions for Continued Airworthiness A320-03444-01-ICA-01, Rev. IR, dated 28/02/2014</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third party.</td>
<td>03/05/2014</td>
<td>Active</td>
<td>10049033</td>
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<td>BABCOCK MISSION CRITICAL SERVICES DESIGNS AND COMpletions LIMITED</td>
<td>REV. 1</td>
<td>AS 355 NP</td>
<td>Installation of a Live Line Working Rig as undertaking load (WL) system for Human External Cargo in accordance with modification BOND/MDM/003 as defined in DL/034-099. Revision 1 is issued to grant extension of the Roblon Rope Life Limitation 9 years and approval for the use of the alternative PCDSB...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- AS335NP Flight Manual Supplement “Live-Line Working Rig”, FMS/034-002 Issue 4 dated 25 Aug 2016 - Instructions for Continued Airworthiness, Part 2 Airworthiness Limitations, ICAM04-018 Issue 5 dated 15 July 2016 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/09/2016</td>
<td>Active</td>
<td>10049035</td>
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<td>SKYLINE HELICOPTERS LTD</td>
<td>REV. 1</td>
<td>429</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Rotorcraft Flight Manual Supplement Longline Hook Control SAS-429, FMS-190, dated 6 June 9th, 2012 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU and EASA.</td>
<td>There is no change to the limitations associated with the cargo hook as a result of longline control circuit installation. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair...</td>
<td>06/05/2014</td>
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<td>S.E.A.M. SERVICE ELECTRONIC AVIATION MARINES A.R.L</td>
<td>REV. 1</td>
<td>GB 35 G52</td>
<td>Installation of a Garmin GTN 650 system, a GA 35 GPS antenna and alhambra/King KN62A DME receiver, according to SEAM modification file no. A320.212 revision 2 dated 15/04/2014</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Garmin Flight Manual Supplement ref.: 190-01007-E3 Rev. 1SEAM Flight Manual Supplement no. SM7-402/12 revision 0 dated23/04/2014 or later revisions of the above listed documents approved by EASA.</td>
<td>The installation covered by this STC is approved for IFR, BRNAV, PRNAV, RNP approach UNNAV and LP approach operations, with coupling to autopilot Bendix/King KAP 150 and to Aspen EFD 1000 Prodisplay System. Installation must be performed by authorized SEAM personnel. Software upgrades introducing new func...</td>
<td>07/05/2014</td>
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<td>ELEDIC LLC</td>
<td>REV. 1</td>
<td>FAA STC2203SE</td>
<td>Installation of a Crane Aerospace &amp; Electronics Tinsensore AssemblyRev. 1 updates the applicability in adding the model BD-700-2A12</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Crane Aerospace Master Document List CC603-022-01 Bombardier , Revision dated 12 September 2013 (+) or later revisions of the above listed document(s) approved/accepted onbehalf EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/06/2021</td>
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<td>10049051</td>
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The Certification Basis (CB) for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment/CS 25 Amrd 28 for the threeparagraphs affected by the change, except for CS 25.1301(c) at Amrd 4. The requirements for enviro...
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<th>EASA STC No.</th>
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<td>10049106</td>
<td>SAFRAN ELECTRONICS &amp; DEFENSES SNS SS - HELICOPTERS DATA AND AIRFRAME NT SOLUTIONS &amp; SERVICES</td>
<td>REV. 1</td>
<td>AS 365 N3</td>
<td>Installation of the HELICOM V2 NUM embedded Equipment for Helicopter AS365 N3+</td>
<td>The change consists of the installation of a Flight Data Monitoring system described in the document capable of acquiring flight datameters to be transmitted and analysed at the end of the flight. Rev.: 1 STC holder name:...</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s): CRI F-01 Li-Ion Battery Installation Dated February 24, 2014. The requirements for the associated technical certification of noise and...</td>
<td>No-AS365.N3+ HV2 NUM- Design Organization Edit A, Rev. 04, Dated May 9th, 2014.N2-AS365.N3+ HV2 N- NUM-Certification Program Edit A, Rev. 04, Dated May 9th, 2014.N3-AS365 N3+ HV2 N- NUM-Design Change Edit A Rev. 04, Dated May 9th, 2014.N7- AS365 N3+ HV2 N- NUM-Installation Manual Edit A Rev. 04, Dated May 9th, 2014.</td>
<td>Prior to installation of this design change it must be determined that the airframe/aircraft/other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/02/2020</td>
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<td>10049107</td>
<td>AEROSPACE RESOURCES, INC.</td>
<td></td>
<td>B.200CGT, B.300C, B.300F, B.200GT 1. B200T, B200CT 1. A200C, A200CT 1. A100-1, 200, 200C, 200T</td>
<td>Approval of Cargo Equipment Installation on Beechcraft KingAir 200 and 300 Series Aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated technical noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Aerospace Resources Inc., FAA Approved Flight Manual Supplement Document No 20-0260-FM65, Revision 0, dated May 29, 2014, or Aerospace Resources...</td>
<td></td>
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<td>Prior to installation of this design change it must be determined that the airframe/aircraft/other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/05/2014</td>
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<td>10049114</td>
<td>DRF STIFTUNG LUFTPFILPFUNG EMMENKUETZIGE</td>
<td></td>
<td>BO105 S, VARIANT: CBS-4</td>
<td>Title: Installation of multiple external and internal mountings for test equipment BO105Description. The mounting for external mission equipment is fixed at the nose of the helicopter. Additional equipment racks are installed in the passenger compartment for various test equipment.</td>
<td>CS 27 Am9, 3 CS 27.1, CS 27.21, CS 27.25, CS 27.27, CS 27.29, CS 27.51, CS 27.71, CS 27.75, CS 27.1411+&lt; CS 27.143, CS 27.173, CS 27.175, CS 27.177, CS 27.241, CS 27.251, CS 27.301, CS 27.3031+&lt; CS 27.305, CS 27.307, CS 27.337, CS 27.471, CS 27.581, CS 27.601, CS 27.603, CS 27.605+&lt; CS 27.607...</td>
<td>Flight Manual Supplement, FAA STC-10-DRF-015 Supplement for Multiplex and internal mountings for test equipment BO105, Ausgabe 00 dated 21st January 2014 or later revisions of the above listed documents approved by EASA-Approval/Flight Manual Supplement Document No 20-0260-FM65, Revision 0, dated May 29, 2014, or Aerospace Resources...</td>
<td></td>
<td></td>
<td>Prior to installation of this design change it must be determined that the airframe/aircraft/other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/05/2014</td>
<td>Active</td>
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<tr>
<td>10049146</td>
<td>IAC - INDUSTRIAS AERONAUTICAS DECOIMIRA, Ltda.</td>
<td></td>
<td>2. 182R, R182, T182, TR1852 2. 182N, 182N, 182P, 182Q, 2. 182R, 182R, 182K, 182L, 2. 182L, 182E, 182F, 182G, 2. 182, 182A, 182B, 182C, 1. 182B, 182T, 182T2</td>
<td>Modification for skydiving operation includes the modification of the right door to open it upwards, theremoval of the copilot control wheel, and the introduction of arrehiment panel protection, a static line parachute support, and an accessory panel fixed to the wheel nut and in the foot...</td>
<td>CS 27 Am9, 3 CS 27.1, CS 27.9, CS 27.21, CS 27.25, CS 27.27, CS 27.29, CS 27.51, CS 27.71, CS 27.75, CS 27.1411+&lt; CS 27.143, CS 27.173, CS 27.175, CS 27.177, CS 27.241, CS 27.251, CS 27.301, CS 27.3031+&lt; CS 27.305, CS 27.307, CS 27.337, CS 27.471, CS 27.581, CS 27.601, CS 27.603, CS 27.605+&lt; CS 27.607...</td>
<td>AFM Supplement For Parachuting Activities IAC-C182-AM5 ED 1or later revisions of the above listed documents approved by EASA-Approval/Flight Manual Supplement Document No 20-0260-FM65, Revision...</td>
<td></td>
<td></td>
<td>Prior to installation of this design change it must be determined that the airframe/aircraft/other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/05/2014</td>
<td>Active</td>
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<tr>
<td>10049148</td>
<td>RUAG AG</td>
<td>REV. 2</td>
<td>EC120B EC 120 B</td>
<td>Installation of Garmin GTN650 Navigation UnitThe initial issue defined the installation of Garmin GTN650 aeronautical equipment for the Garmin GNS430 or GNS430W Revision 1 additionally considers the installation of Garmin GTN650 aeronautical equipment for the Triton 2000 GPS and the Bendix/King KV 165 NAV/COM system.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment CS27 Am9 3 paragraph: 27.1 27.21 27.29 27.351 27.301 27.303 27.305 27.307 27.321 27.337 27.341 27.561 27.601 27.603...</td>
<td>Flight Manual Supplement, doc. ref. FMS-66861 Rev. H.dbl. 19/11/2014Master Data List, doc. ref. MLD-006012, Rev. P.dbl. 19/11/2014Certification Compliance Sheet, doc. ref. CCS-0060191, Rev. P. dbl. 19/11/2014 Certification Compliance Sheet, doc. ref. ICA-0801438, Rev. Cdtbl. 11/11/2014 or later.</td>
<td></td>
<td></td>
<td>GPX uses limited to VFR/GPS approaches prohibited/obstacle and fuel data only for information existing installation of Garmin GNS430/ GNS430W or of Triton 2000 GPSand Bendix/King KV 165 NAV/COM equipment is a prerequisite for this STC. Prior installation of this design change it must be determined...</td>
<td>18/01/2021</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>STC ref.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AIRQUIF A/S</td>
<td></td>
<td>AS 332 L, AS 332 L1</td>
<td>Installation of NVIS Kit in accordance with Airbus MOD DO-332-33-101. The kit comprises installation of cockpit, external and cabin lights, with relative NVIS Light Control Panel as well as cockpit filters.</td>
<td>Certification basis for the original product as amendably thloswing additional or alternative airworthiness requirements - FAR 29 with amendments 26-1 to 26-16. Additional guidance material: AC 29-2C MS16 &amp; RTCA DO-275. The requirements for environmental protection and the associated certificate.</td>
<td>Airbus Flight Manual Supplement Part 1: FMS-332-33-101. Rev. 1 dated 28/02/2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Installation of NVIS Kit does not constitute an approval for NI Operations. Such approval requires coordination with the appropriated operations regulatory authority. Rotorcraft modified by this STC employing Night Vision Imaging System (NVIS) compatible lighting, commonly named NVIS lighting, that has...</td>
<td>19/05/2024</td>
<td>Active</td>
<td>10049153</td>
<td></td>
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<tr>
<td>ROCKWELL</td>
<td>REV. 1</td>
<td>FAA STC SA16769/W</td>
<td>Modification of a Rockwell Collins, Inc. File Server Unit (FSU)/Installation.</td>
<td>Certification basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC. Unfortunately, for environmental protection and the associated certificate(s) and/or emissions levels of the original product, a new configuration of 'Installation of NVIS Kit' is necessary.</td>
<td>Rockwell Collins, Inc. Drawing Index RCA-042-001, Rev. (1) dated 17 February 2012, and later FAA approved revisions, as listed in the attached FAA Approved Model List (AML).</td>
<td>Prior approval installation of a Rockwell Collins, Inc. File Server Unit (FSU) is a prerequisite to this STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>22/11/2018</td>
<td>Active</td>
<td>10049177</td>
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<tr>
<td>AERO DYNAMIX, Inc.</td>
<td>REV. 1</td>
<td>FAA STC SR0965RC</td>
<td>Installation of an NVIS/NVG compatible interior lighting system in accordance with Aero Dynamix, Inc, Master Drawing List MDL 1845-01 Rev B dated 30/04/2014/Rev. 1: Minor Change to STC adding a new configuration of ‘Installation of Night Vision Goggle Compatible Interior Lighting System’ by RFS(Apenn...</td>
<td>Certification basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate(s) and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Drawing List, MDL 1845-01, Rev. C, dated 28 February 2018 - EASA Minor Change Classification EF415-20-105 (1-2HEL) dated 28 February 2020 - Aero Dynamix Project Notification Letter to FAA, dated 5 February 2020 with enclosed Design Change Proposal (8 pages), dated 9 February 2020 - FAA Fo...</td>
<td>Prior to installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>27/02/2020</td>
<td>Active</td>
<td>10049194</td>
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<tr>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>REV. 2</td>
<td>B737-800</td>
<td>B737NG IFE Power Control / EASA DOA EASA.21.1-452 major change/RTCA/DO-465 - Activation of an IFE power isolation switch installed in the P5 overhead panel installed per design change ERT/CNG/523. This retrofit allows the crew to isolate power to the IFE system during emergency conditions and provides a means...</td>
<td>The Certification Basis for the original product as amendably thloswing additional or alternative airworthiness requirements CS-25 Amendment 13, 25.1(a), 25.6(b), 25.601, 25.603 (a), 25.603(c), 25.603(c), 25.611 (a), 25.611(b), 25.130/1(a), 25.130/1(b), 25.130/1(a)(i), 25.130/1(b)(ii), 25.132/2(a)(i),...</td>
<td>Eirtech Aviation Services Ltd Change ERT/465, Master Document List/MDL/465-009 Issue 5, Service Bulletin ERT SB-415 Issue 3, ERT-SB-533 Issue 2, FCOM Supplement FC005W/465-005 Issue 4, Maintenance Manual Supplement MMS/465-012 Issue 2, Maintenance Manual Supplement MMS/465-013 Issue 2...</td>
<td>Applicable to B737-800 with Rockwell Collins IFE STC SA5135NM installed and B737-600/700/800/800-89N-05F</td>
<td>18/05/2015</td>
<td>Active</td>
<td>10049196</td>
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<tr>
<td>A &amp; C PRODUCTS</td>
<td>FAA STC 737-700</td>
<td>Installation of motorized window shades</td>
<td>The Certification Basis for the original product as amendably thloswing additional or alternative airworthiness requirements JAR 25 change 15: 25.551, 25.611, 25.863, 25.130/1(a)(i) for environmental protection and the associated certificate(s) and/or emissions levels of the original product.</td>
<td>Master Data List 737-100-01 DL, Revision E3 for Continued Airworthiness document number ACJ7131I; Revision D for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/49/EC (or subsequent revisions off this decision) and/or the Technical Impac...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>19/05/2014</td>
<td>Active</td>
<td>10049198</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Foreign/NAALC</td>
<td>Movie</td>
<td>Date of Active</td>
<td>Status</td>
<td>System Record</td>
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<td>10049203</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>Modifications of the Unit Loads Devices Limitations/Conditions</td>
<td>FAA STC No. 10049203 FAA STC dated 01 July 2011, 3S Certification, Master Data List (MDL) 058-2325-L01-00, Revision R, dated 30 June 2011, 3S Certification, Aircraft Flight Manual Supplement (AFMS) 058-2325-A01-00, Revision R1, dated 30 June 2011 or later revisions of the above listed documents approved by EASA</td>
<td>FAA STC No. 10049203</td>
<td>CL-600-2B16</td>
<td>03/02/2015</td>
<td>Active</td>
<td>10049203</td>
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<td>10049204</td>
<td>FEDEX EXPRESS CORP.</td>
<td>Modifications of the Unit Loads Devices Limitations/Conditions</td>
<td>B727-100C, B727-200F; Doe Howard Co. Master Drawing Lists 96-001 and 90-002 dated 16 July 13, 1988, B727-200F; Doe Howard Co. Master Drawing Lists 90-001, 90-002 and 90-005 dated 16 July 13, 1998, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent requirements)</td>
<td>FAA STC No. 10049203</td>
<td>727-100C, 727-200F</td>
<td>20/05/2014</td>
<td>Active</td>
<td>10049204</td>
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<td>10049206</td>
<td>FEDEX EXPRESS CORP.</td>
<td>Installation of Expanded Parameter Flight Data Recorder Limitations/Conditions</td>
<td>Top Drawing number B27-31-081 Rev F dated 12 Sept. 2000, Weight and balance supplement B27-31-082 dated 15 June 1998, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementers.</td>
<td>FAA STC No. 10049203</td>
<td>727-100C, 727-200F</td>
<td>10/06/2014</td>
<td>Active</td>
<td>10049206</td>
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<td>10049212</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>Cabin Interior Modification Limitations/Conditions</td>
<td>CS29 Amdt 37e requirements for environmental protection and the associated certificated noise and emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>AS365 N3</td>
<td>MOD 13231 - CVFDR installation</td>
<td>20/05/2014</td>
<td>Active</td>
<td>10049212</td>
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<td>10049225</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>Cabin Interior Modification Limitations/Conditions</td>
<td>Master Data List No. A200-30, FAA STC No. 10049225 FAA STC dated 01 July 2011, 3S Certification, Master Data List (MDL) 058-2325-L01-00, Revision R, dated 30 June 2011, 3S Certification, Aircraft Flight Manual Supplement (AFMS) 058-2325-A01-00, Revision R1, dated 30 June 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent requirements) and/or the Technical Implementers.</td>
<td>FAA STC No. 10049225</td>
<td>A320-212</td>
<td>21/05/2014</td>
<td>Active</td>
<td>10049225</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10049248</td>
<td>INFLIGHT WARNING SYSTEMS, Inc.</td>
<td>FAA STC</td>
<td>ST01820LA</td>
<td>747-400</td>
<td>Installation of a Fan Vibration Monitor Unit (VMU). Installation of Air Conditioning Fan Monitors to monitor bearing performance for the prevention of smoke/odor/fumes into the aircraft cabin and flight deck.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List: Master Drawing List number 13368, Rev A.</td>
<td>Master Drawing List document number 212000, Rev A.</td>
<td>Instructions for Continued Airworthiness (ICA), document number 13368. Rev B or later revisions of the above listed documents approved by EASA.</td>
<td>22/05/2014</td>
<td>Active</td>
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<tr>
<td>10049249</td>
<td>ARMSTRONG AEROSPACE Inc.</td>
<td>FAA STC</td>
<td>ST03284CH</td>
<td>A330-200, A330-300</td>
<td>Certification of Compatibility with Cellular and Wireless Data Technologies. Refer also to Enmoy Aerospace Certification Plan, Document: E1384-2330-C001 rev -</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: 25.1309(a) at CS amendment 225.1529 at CS amendment 27. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data LIST: DC10-9377-00 rev. - (including the references to the documents: Aircraft Equipment List Ref DC10-9377-01 rev. - and Ref DC10-9377-02 rev. -) Certification Plan Document: E1384-2330-C001 rev. - FAA Letter. FAA Statement of Compliance, dated on 16th May 2014 or later revisions of L.</td>
<td>22/05/2014</td>
<td>Active</td>
<td>10140249</td>
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<tr>
<td>10049255</td>
<td>JADE ENGINEERING PTE. LTD.</td>
<td></td>
<td></td>
<td>777-200</td>
<td>BL65 Seat Track installation and Floor Panel Modification</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: 25.1309(a) at Amst 1425.625(a)(b)(c) at Amst 1425.853 at Amst 16. The requirements for environmental protection and the associated certified noise and/or emissions levels remain unchanged.</td>
<td>Master Drawing List JD-SAMC02-1 Rev. NEW, dated 15 May 2014. Declaration of Compliance Form: JD-A/F028, dated 22 May 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>This STC replaces certain floor panel aluminum splice plates with new splice plates made of 17-7PH Ti0105 stainless steel and is approved as an Alternate Method of Compliance (AMOC) to paragraph (f) of EASA's adopted FAA Airworthiness Directive (AD). 2005-18-10 for those splice plates affected by this STC.</td>
<td>23/06/2016</td>
<td>Active</td>
<td>10140255</td>
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<tr>
<td>10049267</td>
<td>AVIONTEC AG</td>
<td>DA-40</td>
<td></td>
<td></td>
<td>Exchange to dual GNS-430W &amp; KN-62A DME Installation: Out: Garmin GNS-430 &amp; GNC-255XL; systems: Dual Garmin GNS-430W &amp; KN-62A systems</td>
<td>The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/approval: CS 23 for the change. The certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List: Master Drawing List No. 3802-017 Rev. 0 of 28.03.2014 for Continued Airworthiness No. 3812-017 Rev. 0 of 02.03.2014 for 3812-017 Initial Issue of 28.03.2014 for later revisions of the above listed documents approved by EASA.</td>
<td>S/N 40043 The approval holder shall fulfill the obligations of Part 21 § 21.111 Prior to installation of this modification it must be determined that the installation is in accordance with the regulations and any other previously installed modifications and or repair will introduce no adverse effects on the...</td>
<td>23/05/2014</td>
<td>Active</td>
<td>10140267</td>
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<td>STC Holder</td>
<td>FAA STC</td>
<td>Rev.</td>
<td>Foreign/NAA Code</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>INFIGHT CANADA Inc.</td>
<td>FAA STC</td>
<td>ST0163WV</td>
<td>TCCA STO-SA14-17 ISSUE 1</td>
<td>737-800</td>
<td>Cabin Recconfiguration installation of first class seats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List: MLD.13167 Rev:1</td>
<td>-Flight Manual Supplement FMS13167 Rev:1</td>
<td>26/05/2014</td>
<td>Active</td>
<td>10040175</td>
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<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>ST0172WV</td>
<td>SAAB SF340A, SAAB 340B</td>
<td>Installation of a Rockwell Collins Attitude Heading Reference System (AHRS) by installing a Rockwell Collins AHC, type AHC-3000A, on SAABSF340A. The Revision 1 modifies the initial applicability by adding the SAAB 340B ..</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Master Drawing List, ref: RCA-0454 Revision B, dated 21 April 2014; -Aircraft Flight Manual Supplement, ref: RCA-0446 Revision B, dated 21 April 2014; -FAA Airliner Flight Manual Supplement, ref: RCA-0453 Revision B, dated 21 April 2014 or later revisions of the above listed documents appr..</td>
<td>None.</td>
<td>14/03/2018</td>
<td>Active</td>
<td>10040200</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
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<td></td>
<td>525, 525A, 526B, 525C</td>
<td>Installation of dual GPS/NAV/COMM Garnir GN5340/530A/W</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23 Amendment 223.307, 365, 561, 575, 603, 607, 611, 615, 671, 867, 1301, 1309, 1321, 1332, 7, 1351, 1357, 1365, 1431, 1529+ (&gt;,&lt;) 1581, 1585 The requirements for environmental protection and t..</td>
<td>Flight Manual Supplement ASR-2011-09-FMS-05-00, Amendment 00 or later revisions of the above listed documents approved by EASA and Instruction for Installation ASR-2011-09-FIL-10-00, Amendment 00-Instructions for Continued Airworthiness ASR-2011-09-ICA-11-00- (&gt;)(&lt;) Amendment 00</td>
<td>None.</td>
<td>27/06/2014</td>
<td>Active</td>
<td>10040071</td>
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<tr>
<td>JETAIR AEROSPAC, LLC</td>
<td>FAA STC</td>
<td>ST00387L-A-D</td>
<td></td>
<td>B727-100/200</td>
<td>Installation of provisions for smoke/fire detection/suppression system in lower cargo compartments. Installation of active smoke/fire detection/suppression system, lower cargo compartments.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Hollingshead Master Data List No. 1361, Rev. C, dated July 11, 1998 (&gt;,&lt;) &gt;Airplane Flight Manual Supplement 2314-04, Rev. 2, dated September 8, 1995 later revisions of the above listed documents approved by EASA and/or other applicable certifying authorities for FAA</td>
<td>Applicability as per above mentioned Master Data List</td>
<td>28/05/2014</td>
<td>Active</td>
<td>10040039</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>STC Date</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10049311</td>
<td>BALLONBAUER, GMBH</td>
<td>03/06/2014</td>
<td>NL-STU940, NL-STU940, NL-STU510, NL-STU1000</td>
<td>Ultralight CT-01 and CT-02 bottom ends beneath NL-STU Series gimbals</td>
<td>CE- or replacement pages for FM: NL-STU, Fahr- und Betriebsanleitung, Satz 5, 5, 10, 12, 13, 16, 20a, Ausgabe Mai 2014. Supplement FM: NL-STU, Fahr- und Betriebsanleitung, Anhang 1, Ultralight Talon Kor, Ausgabe Mai 2014; replacement pages for FM: NL-STU, Flight and Instruction Manual pages 4, 5.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/05/2014</td>
<td>Active</td>
<td>10400401</td>
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<tr>
<td>10049320</td>
<td>AVIATION COMPLETION INNOVATIONS INC</td>
<td>REV. 1</td>
<td>FAA STC SR03007NY</td>
<td>- Configuration 1: Installation of two Class 2 Electronic Flight Bag (EFB) mounting device arm-mounted. Configuration 2: Installation of two Apple iPad or iPad 2 system</td>
<td>- Configuration 1: Air Concepts International, Inc. Installation Drawing List Nos. 41240-00-333-01 and 41240-00-333-04. Revision IR, dated 10 July 2011. - Configuration 2: Air Concepts International, Inc. Installation Drawing List No. 410139-06. Revision IR, dated 11 February 2011. - Modified Aircraft.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/07/2019</td>
<td>Active</td>
<td>10404030</td>
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<tr>
<td>10049362</td>
<td>H4 AEROSPACE (UK) LTD</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>B767-200</td>
<td>20 Bed / 72C Passenger Seat Configuration</td>
<td>None.</td>
<td>None.</td>
<td>02/06/2014</td>
<td>Active</td>
<td>10404062</td>
<td></td>
</tr>
<tr>
<td>10049367</td>
<td>HARTZELL PROPELLER INC</td>
<td>REV. 1</td>
<td>FAA STC SA0414CH-D</td>
<td>M20E</td>
<td>Installation of the Hartzell propeller model HC-C2YR-18BPFT7497 in accordance with Hartzell STC Installation Instructions INST_011007.</td>
<td>Installation of the Hartzell propeller model HC-C2YR-18BPFT7497 in accordance with Hartzell STC Installation Instructions INST_011007.</td>
<td>04/04/2016</td>
<td>Active</td>
<td>10404067</td>
<td></td>
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<tr>
<td>10049374</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td></td>
<td>B737-700</td>
<td>Replacement of two 42&quot; Monitors.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Limited to Model M20E. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/06/2014</td>
<td>Active</td>
<td>10404074</td>
<td></td>
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<tr>
<td>10049398</td>
<td>ELECTRONIC CABLE SPECIALISTS, INC</td>
<td></td>
<td>FAA STC ST03224CH</td>
<td>737-700-800-900-900ER</td>
<td>Installation of ECS EZMount® Trolley Cradle assemblies on existing EZMount® Pivot installations; configuration 1: Cradle (8&quot; x 10&quot;) provisions and configuration 2: Cradle (8.5&quot; x 11.2&quot;) provisions</td>
<td>The Certification Basis for the original product as amended by this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product a...</td>
<td>- The equipment for which these provisions are intended has been bee...</td>
<td>04/04/2016</td>
<td>Active</td>
<td>10404098</td>
<td></td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10049401</td>
<td>PIY ENGINEERING</td>
<td></td>
<td></td>
<td>757-200/-200PF/-300</td>
<td>Electronic Flight Bag (EFB) class 2 installation on Boeing 757-200. This modification deals with the installation of a dual (Captain and First Officer) class 2 Electronic Flight Bag (EFB) system from CMDS/EliteLine, on Boeing 757 Aircraft. Forehead side, this system is composed of an arm mount located on the...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Document List, ref: AVI-00471-MDL-R02, dated 03 June 2014</td>
<td>1. Operations: Refer to §2 of Aircraft Flight Manual Supplement, ref: AVI-00471-AMFS-R02. Airworthiness Limitations: Refer to §6 of Instruction for Continued Airworthiness, ref: AVI-00471-ICA-R00. Prior to installation of this design change it must be determined that the...</td>
<td>04/04/2014</td>
<td>Active</td>
<td>10049401</td>
</tr>
<tr>
<td>10049407</td>
<td>AERODISA S.L.</td>
<td></td>
<td></td>
<td>737-300/-400/-500</td>
<td>Installation of GE AVIATION (SMITHS AEROSPACE - CHELTENHAM) Serviced Mach/Anspaced Indicators (SMASI).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Data List AED2014-10M_MDL rev A2, Supplement to FCOM AED2014-10M_FCOMS Rev A0</td>
<td>Prior to installation of this design change it must be determined that the...</td>
<td>04/04/2014</td>
<td>Active</td>
<td>10049407</td>
</tr>
<tr>
<td>10049411</td>
<td>JET AVIATION AG</td>
<td>REV. 2</td>
<td></td>
<td>737-700</td>
<td>Interior ReconfigurationA The following items are removed from the forward RH lounge area 1: 1. seat cover (STA 480). 2. Side Ledge (STA 495). Coffee table and integrated emergency floor lighting (STA 500CA). Decor panels from the RH bulkhead...</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Data List, Doc. No. JBSL-M-00084MDL-01, Issue 03 Aircraft Flight Manual Supplement, Doc. No. JBSL-M-00084AFM-01, Issue 03 Aircraft Flight Manual Supplement, Doc. No. JBSL-M-00084AFM-02, Issue 03 Aircraft Flight Manual Supplement, Doc. No. JBSL-M-00084ICA-01, Issue 02r</td>
<td>The following modifications are prerequisite to this...</td>
<td>05/07/2021</td>
<td>Active</td>
<td>10049411</td>
</tr>
<tr>
<td>10049432</td>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td>A321-213/-212/-214/-215/-216</td>
<td>LATVISION Cockpit Door Surveillance System (CDS)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>-Master Data List, Aerosol document No. 0709-01-A-301, Revision 0, dated Mar 2014-Instructions for Continued Airworthiness, Aerosol document No. 0709-01-A-221, Revision 0, dated 13 March 2014-Aircraft Flight Manual Supplement, Aerosol document No. 0709-01-A-2301-1, Revision 1, dated 27 May 20...</td>
<td>Prior to installation of this design change it must be determined that the...</td>
<td>06/06/2014</td>
<td>Active</td>
<td>10049432</td>
<td></td>
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<td>10049448</td>
<td>AVIATION COMPLETION INNOVATIONS INC</td>
<td>REV. 1</td>
<td>FAA STC SR02753NY</td>
<td>AB139, AW139</td>
<td>Installation of SAGE Avionics, Inc. ICDS-10 Multi Functions Display</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval, except where amended by additional or later amendments if indicated on FAA STC SR02753NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>- Air Concepts International Inc., Installation Drawing No. 41006-103-00-INS, Revision IR, dated 8 November 2007 - Air Concepts International Inc., Instructions for Continued Airworthiness, Report No. JBCS003, Revision IR, dated 15 November 2009, FAA accepted 12 March 2010- Air Concepts Inter...</td>
<td>Prior to installation of this design change it must be determined that the...</td>
<td>01/07/2019</td>
<td>Active</td>
<td>10049448</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10049449</td>
<td>ROCKWELL</td>
<td>3-</td>
<td>FAA STD SA1684W1</td>
<td>P-180,</td>
<td>Installation of Rockwell Collins, Inc. Aircraft Information Management (AIM) System in defined Part 23 Aircraft.</td>
<td>Certification basis as defined in FAA STD SA1684W1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>To be amended as defined in documents: RCAA - 0408 rev.A (for Cessna models); RCAA - 0409 rev.A (for Beechcraft B200 and B200C), - RCA - 0388 rev.A (for Beechcraft B300 and B300C), - RCA - 0405 rev.A (for Piaggio P. 180). No AFM Supplements have been produced for this STC to be maintained in accord...</td>
<td>09/06/2014</td>
<td>Active</td>
<td>10049449</td>
<td></td>
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<tr>
<td>10049460</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>2-</td>
<td>FAA STC SR02235E</td>
<td>209L-1, 209L-3, 206L-4</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>- Airworthiness Compliance Report DAR-2119-827-1, Rev. C-Master Document List MOL-D-2119-827, Rev. F-Installation Instructions IIN-D119-827, Rev. D-Instructions for Continued Airworthiness ICA. D119-827, Rev. 2 &amp; BCA: D407-827, Rev. 0 prior to installation of this design change.</td>
<td>None</td>
<td>10/06/2014</td>
<td>Active</td>
<td>10049460</td>
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<tr>
<td>10049462</td>
<td>AVONIK STRAUBING ENTWICKLUNGS GMBH</td>
<td>S-76A, S-76B, S-76C</td>
<td>Removal of Flight Management System Universal UNS-1A and Installation of NAA Flight Management System Universal UNS-1x.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement ASR-2010-59-FMS-05-01, dated 01/04/14 &amp; later revisions of the above listed documents approved by EASA Master Document List ASR-2010-59-FMS-05-01, dated 01/04/14</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/06/2014</td>
<td>Active</td>
<td>10049462</td>
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<td>10049468</td>
<td>EICHELDOERFER GmbHFLUGZEUG BAU</td>
<td>AS K 13</td>
<td>Optional Erhöhung der maximalen Abflugmasse / Optional increase of Maximum Mass</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>- Beschreibung des Verfahren zur Erhöhung der Max. Abflugmasse / Description of the procedure for the increase of the max. mass-Handbuch zur Überarbeitung des Flug- und Betriebshandbuchs für das Segelflugzeug AS-K 13 / Handwritten Revision of the Flight and Service Manual for the Sailplane AS-K 13.</td>
<td>Limitations: The increase of the max. mass is limited by 1) The max. mass of the non-lifting parts defined in the TCD-2, increase of the max. 5% of the MTOW due to the difference of the requirements BVS and LFS. Conditions: The sailplane must be in good technical conditions and full-throttle requirement o...</td>
<td>11/06/2014</td>
<td>Active</td>
<td>10049468</td>
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<td>10049473</td>
<td>NSE INDUSTRIES S.A.</td>
<td>REV. 3</td>
<td>AS335F1, AS335F2, AS335W</td>
<td>AS335 NVS Lighting System Installation of NVS Lighting System P/N 45A-1000-V02, defined in MOL-D-45A-1000-V02, or P/N 45B-1000-V02, defined in MOL-D-45B-1000-V02 (with Nigh Vision Goggles.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement RFS-45A-1000-V02 Rev. 02 &amp; later revisions of the above listed documents approved by EASA -AS335 NVS Lighting System Certification Plan - PCERT-45A-1000-V02Rev. 05 - AS335 NVS STC Data List - SDL-45A-1000-V02 Rev. 05 - AS335 VA NAA Master Drawing List - MOL-D4...</td>
<td>The first installation of this kit must be supervised by the STC holder. The rotorcraft employs Night Vision Imaging System lighting (NVIS) that has been certified to ensure the aircraft is compatible with nightvision goggles (NVG). For threaseason modifications that add or change systems that emit on...</td>
<td>22/12/2014</td>
<td>Active</td>
<td>10049473</td>
<td></td>
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<tr>
<td>10049475</td>
<td>ICELANDAIR EHF</td>
<td>757-200</td>
<td>Navigation - Independent Position Determining - Software Update for Replacement of the ACCESS TCAS 2000 Computer to Comply with the TCAS/Change 7.1 Requirements</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>- Change/Repair Approval, Icelandic document No. CAA-757-3440-59(Revision 8N)(-&gt;) dated 3 June 2014 Continued Airworthiness Instructions, Icelandic document No. CAA-757-3440-59(-&gt;) =&gt; Revision A, dated 3 June 2014 Aircraft Flight Manual Supplement, Icelandic document No. AFM-757-3440-59(-&gt;) =&gt; Rev.</td>
<td>None</td>
<td>12/06/2014</td>
<td>Active</td>
<td>10049475</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10049476</td>
<td>SAFRAN SEATS</td>
<td>REV. 1</td>
<td></td>
<td>A330-343</td>
<td>Modification to the aircraft seat installed on Airbus A330-343</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 6, paragraphs various.</td>
<td>Master Data Unit C1145MDL, Issue 1 dated 20 May 2014 or later revisions of the above listed documents approved by EASA</td>
<td>Aircraft Serial Numbers: 1156, 1201, 1206, 1215, 1296, 1315, 1329, 1341 and 1362</td>
<td>12/02/2019</td>
<td>Active</td>
<td>10049476</td>
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<tr>
<td>10049479</td>
<td>H4 AEROSPACE (UK) LTD.</td>
<td>REV. 3</td>
<td>ALL</td>
<td>A330-200</td>
<td>Installation of LED washlights</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>H4 Aerospace Modification Summary and Declaration of Compliance No. MOD444A1705 Issue 10, dated 21 June 2017</td>
<td>Prior to the installation of the new LED Wash Lights on aircraft equipped with a photoluminescent floor path marking system, an analysis is required to be performed considering the impact of thermalisation on the charging capabilities of the floor path marking system.</td>
<td>11/07/2017</td>
<td>Active</td>
<td>10049479</td>
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<tr>
<td>10049481</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td></td>
<td>A330-223</td>
<td>A330-223 new modul pas seats and LD-MCR1: exchange of 285 pas 2-class seating with 275 pas 3-class seating. Newest models Thompson Vantage (22 BC) and 2M Eco 01 (252 YCC). Integration of Lower Deck Modular Crew Rest (LD-MCR) containing staircase and secondary escape path (hash).3. Assis...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document STO225-13/33M Issue2 rev 0-ICA, ref A-00-14/16-ICA, Issue 1 rev 1. Compliance/Record List CRULS-13/33M Issue 2 rev 0. Latest revision of the above listed documents approved by EASA</td>
<td>1) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change or/ and repair will introduce roadworthy eff...</td>
<td>16/07/2014</td>
<td>Active</td>
<td>10049481</td>
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<tr>
<td>10049486</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 3</td>
<td>777-200</td>
<td>777-200</td>
<td>B772 BEST Cabin reconfiguration (Installation of Zodiac SeaFrance Business class (Cirrus) and Economy class seats (Zom)). Modification of B/E Aerospace Premium Economic class seats. Installation of new partitions and stowage as well as airconditioning of existing galleys, lavatories (&lt;,&gt;). Assist...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s)-CRJ D-02 (Single Side Facing Seats)-CRJ D-03 (Seats with Inflatable Restraints) (The requirements for environmental protection and the associated certifi...</td>
<td>- Change Approval Sheet, Air France document No. CAAS-777-12-008 Revision 0(,)&gt; dated 11 June 2014. Aircraft Maintenance Manual Supplement, Air France document No. AAMIS-777-12-008-001, Revision 0, dated 1 April 2014 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>13/06/2014</td>
<td>Active</td>
<td>10049486</td>
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<tr>
<td>10049498</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 2</td>
<td>FALCON 10</td>
<td>EGIPS Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-ACNS Initial Issue, Subpart E, Section 1, TAW97 The requirements for environmental protection and the associated certified noise and or emissions levels of the original p...</td>
<td>AFM Supplement ref. AE14039AFMS Issue 0, dated 23/04/2014. Master Data List ref. AE14039MDL Issue 2, dated 11/06/2014 orater revisions of the above listed documents approved by EASA</td>
<td>Applicable to Falcon 10 model, S/N 155</td>
<td>13/06/2014</td>
<td>Active</td>
<td>10049498</td>
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<tr>
<td>10049505</td>
<td>ULMAT ENGINEERING GMBH</td>
<td>REV. 2</td>
<td>A330-223 A330-200</td>
<td>A330-223 Cabin Layout Conversion The change contains a modification to a A330 cabin after selling to a new operator. The change comprises: Seat placement, floor covering, PSU Layout, Bar removal, Relocation of overhead bins, Stowage’s and Lining/Configuration...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Configuration 1 P3 Voith Aerospace Master Data List PSV-100021-MDL-0001-A, Initial Issue/Configuration 2 Umlaut Master Document List UE-AAA01-MDL-0001-Umlaut Aircraft Flight Manual Supplement (AFMS) UE-AAA...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change or/ and repair will introduce roadworthy eff...</td>
<td>19/01/2021</td>
<td>Active</td>
<td>10049505</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10049507</td>
<td>AERODATA AG</td>
<td>B300</td>
<td>Installation of Aerodata Advanced Flight Inspection System &quot;AD-AFIS-0113&quot;.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23 Amdt. 3 elected to comply with, except 23.561 and 23.785 (kept FAR23.561 Amdt. 34 and FAR 23.785 Amdt. 34), i.e. the requirements for environmental protection and the associated certification noise and/or emissions levels of the original product unamended and remain applicable to the certificate/approval.</td>
<td>Installation of Aerodata Advanced Flight Inspection System &quot;AD-AFIS-0113&quot; in accordance with Aerodata document MDL-4/C-085, revision 0, dated March 25, 2014 aircraft Flight Manual, Supplement, Aerodata document AFM-143, for maintenance of the above listed documents approved by EASA.</td>
<td>16/06/2014</td>
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<td>10049507</td>
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<td>10049522</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>PA 46-310P</td>
<td>MOD 13120 - Installation PFD EFD1000/PRO/STN650</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated-certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>AE13120ICA Rev. 0 dated 28 Aug 2014 AEROTEC A319/320/MD-11F Master Data List Rev. 1 dated 17 Oct 2013 AEROTEC A319/320/MD-11F Master Data List Rev. 1 dated 28 Aug 2013 AEROTEC A319/320/MD-11F Master Data List Rev. 1 dated 28 Aug 2013 AEROTEC A319/320/MD-11F Master Data List Rev. 1 dated 28 Aug 2013</td>
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<td>10049524</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 5</td>
<td>A321-211, A321-231, A320-211, A321-111, A321-211, A320-211, A320-216, A319-112, A319-114, A319-112</td>
<td>Activation of Wireless IFE system Board Connect.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated-certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>Configuration 1: - Supplemental Type Definition Document STDD23-1232M Issue 1 Rev. 2 dated 07/08/2014. - Compliance Record List CRLL23-1323M Issue 2 Rev. 2 dated 11/03/2015. - Instruction for Continued Airworthiness A-06-1400-ICA Issue Rev. 1 dated 08/06/2014. - A319 Flight Manual.</td>
<td>28/11/2017</td>
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<td>10049566</td>
<td>AAR CARGO SYSTEMS</td>
<td>FAA STC ST01859CH</td>
<td>MD-11F</td>
<td>Main Deck Cargo Handling System (FAA STC No. ST01859CH)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated-certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>AAR Report 1562 Revision A, July 1st, 2003 - Drawing List Report AAR Report 1568 Issue 2, November 5th, 2013 - Instructions for Continued Airworthiness AAR Report 1991 Revision - May 1st, 2003 - Weight and Balance Reporters later revisions of the above listed documents approved by EASA</td>
<td>20/06/2014</td>
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<td>STC Holder</td>
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<td>Dunlop Aircraft Tyres Limited</td>
<td>CL-600-2E25 (CRJ-1000)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Dunlop Aircraft Tyres Master Document List No. DATL, PR 10 MDL, Rev. A dated 12 May 2014</td>
<td>No further revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>20/06/2014</td>
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<td>10049572</td>
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<td>AIM Aviation (Henshalls) Ltd</td>
<td>A321-231, A321-232, A321-211, A321-212, A321-213</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AIM Aviation (Henshalls) List Modification No. HEN 1280 Issue 1 for later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>20/06/2014</td>
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<td>Northern Avionics S.R.L.</td>
<td>AW109SP</td>
<td>NAV 483 &quot;Installation of Helicopter Emergency Egress Light (HEEL) System&quot;</td>
<td>RFMS NAV-483, Rev. 0 for later revisions of the above listed documents approved by EASA</td>
<td>Modification Bulletin NAV-483, Rev. 0, Installation Instructions II-NAV-483-180, Part A, Installation Instructions II-NAV-483-280, Part B Installation for Continuous Airworthiness ICA NAV-483, Rev. 0</td>
<td>None</td>
<td>20/06/2014</td>
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<td>S4A Solutions For Aviation, S.L.</td>
<td>F 172L, F 172M, F 172N, 172L, 172M, 172N</td>
<td>Installation of banner towing hook TOST EB5</td>
<td>- Installation in accordance with Boletín de Servicio, ref. 17201-S0801, dated October 30, 2013 - Operation in accordance with Flight Manual Supplement, ref. 17201-HD001, edition 4, dated 5 June 2014 - Instructions for Continued Airworthiness in accordance with Supplement 1 manual de mantenimiento...</td>
<td>As per the limitations section of the Flight Manual Supplement, ref. 17201-HD001, edition 4, dated 5 June 2014... Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will...</td>
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<td>Beechcraft Corporation</td>
<td>400A</td>
<td>Installation of an Increased Ampere-Hour Recombinant Gas (RG) Lead Acid Battery.</td>
<td>Installation in accordance with the Flight Manual Supplement, ref. 17201-HD001, dated September 5, 2001, Instructions for Continued Airworthiness in accordance with the Flight Manual Supplement, ref. 17201-HD001, edition 4, dated 5 June 2014... Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will...</td>
<td>Aircraft Serial RK-98 and subsequent Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will...</td>
<td>23/06/2014</td>
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<td>Avionik Straubing Entwicklungs GmbH</td>
<td>EC 135 T2, EC 135 T2+ EC 135 P2+ EC 135 T1 EC 135 P1, EC 135 P2</td>
<td>Installation of Garmin GTN650 and GTN750</td>
<td>Rotorcraft Flight Manual Supplement ASR-2012-043- FMS-05-00, Dated 17/06/2014, 14th for later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>24/06/2014</td>
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<td>10046616</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>737-800, B300C, B300, B300C, 200, 300, 300LU</td>
<td>Installation of Dual GARMIN GTX 33D Transponder System for Mode S ELTs</td>
<td>EASA approved Engineering Order, Doc. No.: 2087300-EO, issued 1, dated 23-May-2014; EASA approved Aircraft Flight Manual Supplement, Doc. No. 2087300-AM(+) dated 23-May-2014; Initial Issue 1, dated 6- Feb-2014;幼儿园/foreword revisions of the above listed documents approved by EASA.</td>
<td>None</td>
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<td>VOYAGEUR AVIATION CORP.</td>
<td>DHC-8-202</td>
<td>TCAS II Version 7.1 / Mode S Transponder Upgrade</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation must be in accordance with TCCA approved TCAS II Certification System and any other previously installed design change and/or repair must introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2014</td>
<td>Active</td>
<td>10049624</td>
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<td>ROEDER PRAEZISION GMBH</td>
<td>A320-232</td>
<td>Galley G5 and Lavatory D installation. (removal of galley G4, lavatory F, partition 14 and cabin attendant seat. Installation of new/refurbished galley G5, Lavatory D and 2 cabin attendant seats. Modification of complete galley entry, area doors for these installations. Modification of structures...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Limited to MSN 2587.</td>
<td>15/06/2018</td>
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<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>A300F4-608ST</td>
<td>Auxiliary Fin Tip reengineering update.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>08/07/2015</td>
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<td>10049630</td>
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<td>SKYTRAC SYSTEMS LTD.</td>
<td>DHC-8-401, DHC-8-402, DHC-8-314, DHC-8-315, DHC-8-301, DHC-8-311, DHC-8-301, DHC-8-202, DHC-8-106, DHC-8-102, DHC-8-103</td>
<td>SkyTrac Flight Following/ SATCOM System Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels...</td>
<td>Installation must be in accordance with TCCA approved TCAS II Certification System and any other previously installed design change and/or repair must introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/06/2014</td>
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<td>10049644</td>
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<td>FOKKER SERVICES B.V.</td>
<td>737-700, 737-800, 737-900</td>
<td>Upgrade TCAS 7.0 (TPA 81A) to 7.1 (TPA 108B) on Boeing 737</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Plan CP-0223 Issue 1/Compliance Record Report CR-0223 Issuance and any later revisions of the above listed documents approved by EASA.</td>
<td>27/06/2014</td>
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<td>EASA STC No.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>1004666</td>
<td>FOKKER SERVICES B.V.</td>
<td>03/07/2024</td>
<td>10049672</td>
<td>CPDLIC (Controller Pilot Data Link Communication) in the Boeing 737</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Parts: CP-0212 Issue 3, CP-213 Issue 1, CP-0219 Issue 1, CP-216 Issue 1</td>
<td>2Compliance Record Reports: CRR-0212 Issue 1, CRR-0213 Issue 1, CRR-0219 Issue 1, CRR-0216 Issue 1</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>1004672</td>
<td>AVIATION COMPLETION INNOVATIONS INC.</td>
<td>REV. 3</td>
<td>FAA STC SR-02325NY</td>
<td>Installation of Cabin Interior Sound Reduction Wall Panels System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SR-02325NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Air Concepts International, LLC Installation Drawing List Report 01/12-314-INS, Revision E, dated 9 March 2010: Configuration 1, 2, &amp; 3; IAW Installation Instructions Report IM000139-02, Revision 8, dated 9 March 2010: Configuration 1 Left Side Sliding Door, Left Side Only - Configuration 2 Right Sid...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2019</td>
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<td>1004865</td>
<td>AERO TWIN, Inc.</td>
<td>FAA ST-0259AK</td>
<td>C-212-CE, C-212-CF, C-212-DJ, C-212-DB, C-212-CC, C-212-CO,</td>
<td>&quot;CASA 212 Exhaust Deflectors&quot; Installation of Aero Twin, Inc. CASA 212 Exhaust Deflectors, part number AT-CASA-EXDF-1001 in accordance with Aero Twin, Inc. Master Data List No. AT-CASA-EXDF-1001, Original Issue, dated June 22, 2010, or later FAA-approved revision. This is an approximate 12&quot; stainless steel panel.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Cap. 2.9 &quot;Airworthiness Limitations&quot; contained in Doc.&quot;Instructions for Continued Airworthiness No. CASA-212-EXDF-1001&quot;, Original Issue, dated June 25, 2010, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this).</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/06/2014</td>
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<td>1004960</td>
<td>VERO BEACH AVIONICS, INC.</td>
<td>FAA ST-0405SAT</td>
<td>PA-46-500TP, PA-46-350P, PA-46-350T</td>
<td>Installation of Garmin GSR 56 System / This STC installs Garmin GSR 56 System which provides Iridium Satellite SATCOM phone service, SMS Text Messaging and World Wide Weather Service. The system is integrated into the Garmin G1000 EFIS and interfaces with the aircraft audio panel.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Maintenance shall be accomplished in accordance with Avionics Support Group Instructions for Continued Airworthiness 212145-ICA, Rev. C, The system should be installed in accordance with SAE: AS2013-035-1, Rev B or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this).</td>
<td>Installation is limited to aircraft with the Garmin G1000 EFIS. This approval should not be extended to other aircraft of this model one which have previously approved modifications that have been installed and are in service, unless is determined by the investigator that the interrelationship between the design change and any other p...</td>
<td>30/06/2014</td>
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<td>1004964</td>
<td>PMV ENGINEERING</td>
<td>FAA ST-0405SAT</td>
<td>767-300, 767-400ER, 767-200, 767-300,</td>
<td>Electronic Flight Bag (EFB) class 2 installation on Boeing 767. This modification deals with the installation of a dual (Captain and/or First Officer) class 2 Electronic Flight Bag (EFB) system from OMDJElectronics, on Boeing 767 Aircraft. For each side, this system is composed of a mount arm located on wind...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Document List, ref.: AVI-00530-MDL-00\textsuperscript{1} The ‘Operation of Aircraft Flight Manual Supplement, ref.: AVI-00530-AMS-R02: Prior to installation of this design change it must be determined that the interrelationship between th...</td>
<td>30/06/2014</td>
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<tr>
<td>NORTWEST AEROSPACE TECHNOLOGIES Inc.</td>
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<td>STC 040311121</td>
<td>A320-213</td>
<td>passenger interior modification on British Airways A320-213 aircraft</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: JAR 25 Change 13 25.562(a)(b) &amp; (c) ; 25.853 (a)(b) &amp; (c). The requirements for environmental protection and the associated Noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>Master Drawing UML Interior Reconfiguration on British Airways A320-213 aircraft, N4460-D10 Rev. B dated 29 June 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>British Airways Service Bulletin SB A320-25-040 must be installed at the same time as this STC. Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/07/2014</td>
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<td>BOMBARDIER INC.</td>
<td>TCCA C-LSA03-341D</td>
<td>BD-700-1A10</td>
<td></td>
<td>Aircraft Completion B 10 passenger interior configuration - EASA validation of Transport Canada STC C-LSA03-354D at issue 1</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Development(s): EASA Deviation CRI D-100 at issue 3 &quot;Doors between passenger compartment&quot; and EASA Deviation CRI D-300 at issue 3 &quot;Passenger information sign&quot;. The requirement...</td>
<td>Modification to be completed in accordance with Boeing Consulting Services Modification Data Summary D24000 Revision A dated 16th June 2014 Required Flight Manual Supplement: G- FM2400030 rev N/C - Cabin master power switch- G-FM2500061 rev N/C - Interior completion- G-FM2510007 rev N/C - Flight crew...</td>
<td>- This STC is limited to aircraft serial number 9110. - Refer to the required Flight Manual Supplements and Amendments, Limitations Supplement dated in associated technical documentation - Bombardier Service Bulletin 700-31-018 (Modification to Integrated electronics Computers (IEC) - Avionics 2001 Bat...</td>
<td>01/07/2014</td>
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<td>BOMBARDIER INC.</td>
<td>TCCA C-LSA03-394D</td>
<td>BD-700-1A10</td>
<td></td>
<td>Modifications to Heads Up Technologies CMS-400 Checklist Management System - EASA validation of Transport Canada STC C-LSA03-394D at issue 1. Modification to be completed in accordance with Aero Consulting Services Modification Data Summary D24000030 revision N/C or later approved revisions.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Development(s): EASA Deviation CRI D-100 at issue 3 &quot;Doors between passenger compartment&quot; and EASA Deviation CRI D-300 at issue 3 &quot;Passenger information sign&quot;. The requirement...</td>
<td>Modification to be completed in accordance with Boeing Consulting Services Modification Data Summary D24000 Revision A dated 16th June 2014 Required Flight Manual Supplement: G- FM2400030 rev N/C - Cabin master power switch- G-FM2500061 rev N/C - Interior completion- G-FM2510007 rev N/C - Flight crew...</td>
<td>- This STC is limited to aircraft serial number 9110. - Refer to the &quot;Limitations&quot; sections of the EASA approved Flight Manual Supplement G- FM2400030 revision A or later approved revisions. - Previous installation of CMS-400 Checklist Management System Inconvenience with Transport Canada...</td>
<td>01/07/2014</td>
<td>Active</td>
<td>10408704</td>
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<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 1</td>
<td>A321-231, A321-232</td>
<td></td>
<td>Activation of Rockwell Collins iPaves IE System Rev 1: Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: JAR 25 Change 11, Para25.56, CS 25 Amendment 4, CS 25.29, CS25.301, CS 25.303, CS 25.305, CS25.307, CS 25.581, CS 25.601.</td>
<td>Modification AES-321-3068, Issue 2 dated 28 June 2014 Relevant Data List AES-321-3068MDL Issue 2 dated 28 June 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/04/2021</td>
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<td>ROCKWELL COLLINS, Inc.</td>
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<td>Modification of a Rockwell Collins, Inc. AHC-85 ) (AHRS) installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>STC ST01736W, Issue October 11, 2013</td>
<td>None.</td>
<td>07/07/2014</td>
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<td>STC-01736W</td>
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<td>SWS CERTIFICATION SERVICES LTD.</td>
<td>REV. 1</td>
<td>ANZ 787-9 Sycoucch Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>SWS Certification Services Ltd Master Configuration Schema; 3106-MCS-01 Issue 1 or later revisions of the above listed documents approved by EASA.</td>
<td>07/07/2014</td>
<td>Active</td>
<td>STC-017672</td>
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<tr>
<td>AVONIK STRABING ENTWICKLUNGS GmbH</td>
<td></td>
<td>Installation of Flight Management System Universal UNS-11/w</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 2 (not required by 21A.101)1 (10)3 (R)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3 (2)4 (2)5 (3)6 (10)7 (2)3</td>
<td>Flight Manual Supplement ASR-2011-064-PMS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA. - Instructions for Installation ASR-2011-064-FMS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA. - Instructions for Installation ASR-2011-064-FMS-05-00 Rev. 00 or later revisions of the above listed documents approved by EASA. - Continued Airworthiness ICA202-1R...</td>
<td>07/07/2014</td>
<td>Active</td>
<td>STC-017672</td>
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<td>ICELANDAIR DESIGN ORGANISATION</td>
<td></td>
<td>Installation of an AVIO Interio business class and ecomomyclass seating interior and Inflight Entertainment System (IFE) for Bell</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List GR00012-MDL rev. 00 Instructions for Continued Airworthiness GR00013-ICA rev. IRAircraft Maintenance Manual Supplement GR00014-AIMM rev. IFweight Balance Report GR00030-WBR rev. 00 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>07/07/2014</td>
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<td>STC-017672</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>AIRLEC AIR ESPACE</td>
<td>501</td>
<td>P-RNAV capability of a Dual Garmin GNS 400/500A nav system</td>
<td>Flight Manual Supplement FRAHFM00124 Revision 0</td>
<td>The Certification Basis for the original product as amended by the following additional or later revisions of the above listed documents approved by EASA. Change Statement DC-0251 Issue 2</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change or/and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/07/2014</td>
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<td>1040405</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>PA-46-300P (MALIBU)</td>
<td>Installation of 5-blade MT-27-1-E-C-F-R (PY) CFR203-58d</td>
<td>MT-Propeller Master Document List No. E-2456To be operated in accordance with MT-Propeller AFM-S Document No. E-2465or later RevisionsTo be maintained as a MT-Propeller ICA Document No. E-2461</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change or/and repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/07/2014</td>
<td>Active</td>
<td>1040411</td>
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<td>E.A.D AEROSPACE AIRWORTHINESS</td>
<td>REV. 2</td>
<td>Installation of High Gain Antenna - EM-157640</td>
<td>Master Document List, EAD Aerospace document No. 01441-EAS-A-2246or later revisions of the above listed documents approved by EASA Master Installation Instructions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change or/and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/06/2019</td>
<td>Active</td>
<td>1040417</td>
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<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 2</td>
<td>Modification of a Rockwell Collins, Inc. File Server Unit (FSU)</td>
<td>Rockwell Collins, Inc. Drawing Index RCA-043-001, Rev. (-) dated 17February 2012, or later FAA approved revision, as listed on the attachedFAA Approved Model List (AML).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the above listed document (s) approved are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/11/2018</td>
<td>Active</td>
<td>1040420</td>
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<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>Installation of Rockwell Collins, Inc. Aircraft Information Management(AIM) System</td>
<td>FAA Approved Model List (AML) RCA-039B, Revision 6 dated 7July 2012, or later FAA approved revision 2</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the above listed document(s) approved are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/10/2018</td>
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<td>1040421</td>
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<td>STC Holder</td>
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<td>FOKKER SERVICES B.V.</td>
<td>10049832</td>
<td>CPDLC (Controller Pilot Data Link Communication) in the Boeing 757</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Parts: CP-0214 Issue 1, CP-220 Issue 2, Compliance Record Reports: CRR-0214 Issue 1, CRR-02120 Issue 15E-782 Issue 1, Boeing B757-GPS antenna ICA Supplement B757-EX5- AFM-5-001 Revision 1: AFM Supplement B757 CPDLC installation later revisions of the above listed documents approved by...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no noxious effect upon the airworthiness of the product.</td>
<td>10/07/2014</td>
<td>Active</td>
<td>10049822</td>
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<td>LUTTHANSA TECHNO AG</td>
<td>10049830</td>
<td>A330-300 new model pax seats and monuments installation 1. Exchange of 332 pax 2-class seating with 315 pax 2-class seating. Nearest seats model Thompson Vantare (BC) and ZIM Eco 01 (Y/C) indistinguishable IFE equipment. 2. Installation of monuments and associated cabin interior modifications</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document STD25-14/02M issue 1 rev 1, ICA, ref A-06-1417-ICA, Issue 1 rev 0- Compliance Record List CR25-14/02M is 1 rev 1 for later revisions of the above listed documents approved by EASA.</td>
<td>1) Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. 2) This modification must be installed in co...</td>
<td>10/07/2014</td>
<td>Active</td>
<td>10049829</td>
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<td>AERO MEDICAL PRODUCTS MPG. Inc.</td>
<td>10049831</td>
<td>FAA STC 5A1291GL</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>Aero Medical Products, Installation &amp; Operating Instructions, EA Form 0156, rev. 1 dated 23 April 2001 in aero Medical Products, Maintenance Requirements for Aircraft, Libras, EA Form 107, rev. 1 dated 05 July 1990 later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>11/07/2014</td>
<td>Active</td>
<td>10049832</td>
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<tr>
<td>PW ENGINEERING GmbH</td>
<td>10049832</td>
<td>REV. 2 B757-200 Cabin Interior Change New LOPA in the economy class, including ETSO seats with embedded LCD and in-seat power supply, new In-Flight Entertainment System for the whole aircraft, including a control monitor unit for the cabin attendants, extra head LCO monitors for passenger addressing, etc...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>- Instructions for Continuous Airworthiness, PFH reference B752A-ICA-Airworthiness MAR 31/2014, Revision 03 dated Dec 01/2014, Aircraft Flight Manual Supplement, PFH reference B752A-AM-MA issue 01 dated March 31, 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>1. This approval is strictly limited to the following B757-200 aircraft Serial No 28112, 29488 and 29489, sharing the same initial cabin lay-out/configuration.2. Operational limitations: refer to section 2 of Aircraft Flight Manual Supplement B752A-AM-A.3. Airworthiness limitations: refer to section B...</td>
<td>13/01/2015</td>
<td>Active</td>
<td>10049832</td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES NC.</td>
<td>10049833</td>
<td>REV. 1 777-200 Interior Recomfiguration for Cathay Pacific Airways Boeing 777-200 Aircraft The design change encompasses an interior recomfiguration on four Boeing 777-200 aircraft having the same initial interior configuration: rewound salvaged/refreshed interior furnishings (galleys, overhead bins, closets, lavatories...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>- Instructions for Continued Airworthiness, NAT reference N046-207-00 rev B-Master Drawing List, NAT reference N046-207-01 rev. E for later revisions of the above listed documents approved by EASA. Certification summary. NAT reference N044-2010rev. C: This approval is strictly limited to the five (5) following 777-274 aircraft Serial No 27116, 27255, 27285, 27295. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair...</td>
<td>15/07/2014</td>
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<td>10049837</td>
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<td>GFD GmbH</td>
<td>10049838</td>
<td>REV. 1 35A Iridium SATCOM Dual Antenna provisionrev. Rev. 1 Transfer to GFD GmbH</td>
<td>FAR 25The Certification Basis for the original product and the following/additional or alternative airworthiness requirements are applicable to this certificate/approval.</td>
<td>- ICA Instructions for Continued Airworthiness Doc: 532-081 Rev A- Master Data List Doc: 532-089 Rev. Arit later revisions of the above listed documents(s) approved: acceptable under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/08/2014</td>
<td>Active</td>
<td>10049839</td>
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An agency of the European Union
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<th>STC Holder</th>
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<th>Models</th>
<th>Description</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>GFD GmbH</td>
<td>10049839</td>
<td>REV. 1</td>
<td>35A, 36A</td>
<td>2nd SM GPS Antenna Provision Rev. 1</td>
<td>Transfer to GFD GmbH</td>
<td>FAR 25-The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to the certificate/approval:</td>
<td>- ICA Inkstructions for Continued Airworthiness Doc. 5269-065 Rev. RI, Master Data List Doc. 5229-049 Rev. B or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it may be determined that the relationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/02/2022</td>
<td>Active</td>
<td>10049839</td>
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<td>PRO STAR AVIATION, L.L.C.</td>
<td>10049852</td>
<td>REV. 1</td>
<td>FAA STC SA00388O</td>
<td>B200, B200C, 300, B300, B300C</td>
<td>Installation of a Thane &amp; Thane AVIATOR 200 Swift Broadband System. As defined in FAA STC ref SA00388O.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Pro Star Aviation Installation Data List No. 2857-DL-101 Rev IR, dated December 06, 2013, Operation in accordance with Pro Star Aviation Airplane Flight Manual Supplement ref. 2857-FMS-101 Rev IR, dated December 07, 2013, Maintain in accordance with Pro Star Aviation Inst...</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/07/2014</td>
<td>Active</td>
<td>10049852</td>
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<td>LATITUDE TECHNOLOGIES CORPORATION</td>
<td>10049855</td>
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<td>TCCA STC 13-35 ISSUE 1</td>
<td>EC 130 B4 AS 350 B, AS 350 B1 AS 350 D AS 350 B2, AS 350 B3</td>
<td>Installation of the Latitude Technology IONode Flight Data Monitoring (IONode Flight Data Monitoring (IONode for Eurocopter AS 350B, B1, B2, B3, BA, D and D1, EC130B4, EC120, 1126-400-002 Rev. NC), Instructions for Continuing Airworthiness for the Installation of IONode Flight Data Monitor Installation Instruction 1126-400-002.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Installation Instructions Latitude Technologies IONode for Eurocopter AS 350B, B1, B2, B3, BA, D and D1, EC130B4, EC120, 1126-400-002 Rev. NC, - Instructions for Continuing Airworthiness for the Installation of Latitude Technologies IONode Flight Data Monitor Installation Instruction 1126-400-002 Rev. NC.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/07/2014</td>
<td>Active</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD</td>
<td>10049863</td>
<td></td>
<td>CAA UK RESPONSIBILITY 8300</td>
<td></td>
<td>Approval of Universal Avionics UMS-1 Life FMS for PNR/NAV Operations.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Operation in accordance with Bournemouth Aviation (Consultants) Ltd Flight Manual Supplement Number BAC-909RM001 Revision 1, dated July 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>14/07/2014</td>
<td>Active</td>
<td>10049863</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>10049864</td>
<td></td>
<td>FAA STC ST0139NY</td>
<td>A330-200 / -300</td>
<td>Panasonic eXLife In-Flight Entertainment System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Refer to section 3 in Compliance Summary Report ref. 6654-08714-2362 rev 4. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- MASTER DATA LIST : 0154-08715-2362, rev A - ICA &amp; EWS for Continued Airworthiness : 0254-08712-2362, rev A - Compliance Summary Report Project results : 6654-08714-2362, rev B - AFM Supplement 0654-08722-2362, rev A - FAA Letter 'EASA Approval of FAA Supplement ref to Type Certificate (STC) ST0139NY', dated 18 April 2014 or later revisions of the above documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/07/2014</td>
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<td>PILATUS BUSINESS AIRCRAFT, LTD.</td>
<td>10049872</td>
<td>REV. 2</td>
<td>FAA STC SA04153AT</td>
<td>PC-12/47E</td>
<td>EMTEQ eConnect In-Flight entertainment System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>To be installed according to document Master Drawing List N.2-125214-001 Rev. A dated 28 April 2014 to be operated according to AFMS 2-125214-800 Rev. RI dated 16 May 2014 to be maintained according to (ICA) engineering document N. N. 2-125214-802 Rev. B dated 18 April 2014 or later revisions of the above...</td>
<td>None.</td>
<td>23/02/2015</td>
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The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: 25.305(a)(b) - 25 Amrd 1425.961 - CS 25 Amrd 1425.813 - CS 25 Amrd 1425.785 - CS 25 Amrd 1425.765.

The Certification Basis (CB) for the original product remains applicable to the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: 25.305(a)(b) - 25 Amrd 1425.961 - CS 25 Amrd 1425.813 - CS 25 Amrd 1425.785 - CS 25 Amrd 1425.765.

To be installed according to FAAapproved March 6, 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/40/EC (or subsequent revisions of this decision) and/or the Technical Implementation Procedure...
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<th>Foreign/NAA STC ref.</th>
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>10049911</td>
<td>747-400</td>
<td></td>
<td>Communications - Installation of AGO Software-update of the ACARS CMU Software</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>CP23-1423M Issue 0, Revision 0 (2014-06-10) - Certification Program DL/L23-1423M, Issue 1, Revision 0 (2014-07-15) - Compliance Record List 00-1422-ICA, Issue 1, Revision 0 (2014-04-28) - Instructions for Continued Airworthiness STDD23-1423 38I, Issue 1, Revision 0 (2014-07-15) - Supplemental Type Def.</td>
<td>None</td>
<td>17/07/2014</td>
<td>Active</td>
<td>10249971</td>
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<tr>
<td>DECA AVIATION ENGINEERING Ltd./AIA PAL AEROSPACE</td>
<td>10049913</td>
<td>O-LSA10-033/D</td>
<td>BD-700-1A10</td>
<td>Indium Antenna Installation (Structural Provisions)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada STC No. O-LSA10-033/D, Issue 2 dated 26 February 2010 lnntech Aviation Engineering Drawing IL/DC-0036-019, Revision NC DECA Aviation Engineering Instructions for Continued Airworthiness CA10019, Revision 4 or later revisions of the above listed documents approved by EASA in accordance with</td>
<td>None</td>
<td>17/07/2014</td>
<td>Active</td>
<td>10249913</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>10049916</td>
<td>767-200, 767-300, 767-300F</td>
<td></td>
<td>Introduction of Dink + WCNC</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRF F-Ch. Data Link Services for the Single European Sky</td>
<td>Flight Manual Supplement S21 FMS-004 Issue 1 for later revisions of the above listed documents approved by EASA Master Data List S21-MD-050 Issue 4</td>
<td>None</td>
<td>18/07/2014</td>
<td>Active</td>
<td>10249916</td>
<td></td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>10049921</td>
<td>757-200</td>
<td></td>
<td>8757 FULL BUSINESS CONFIGURATION Modification of the cabin to achieve a full business configuration (TAPAX), with IFE/Moving Map system, In-Seat Power Supply, and 115V/60Hz inho cockpit.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-01 &quot;Application of healthrisk and smoke density requirements tosaat material&quot;, Issue 1 dated 03-06-2014. The requirements for environmental pr...</td>
<td>- Airplane Flight Manual Supplement. AEROTEC document number AE14028A/MS Issue 0 dated 08/08/2014 - Instructions for Continued Airworthiness. AEROTEC document number AE14028B/C Issue 0 dated 30/04/2014 or later revisions of the above listed documents approved by EASA Master Data List, AEROTEC document...</td>
<td>21/07/2014</td>
<td>Active</td>
<td>10249921</td>
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<tr>
<td>MARSHALL OF CAMBRIDGE AEROSPACELIMITED</td>
<td>10049937</td>
<td>AA330-200</td>
<td>A330 Cabin and In Flight Entertainment Upgrade including partition of Lower Deck Mobile Crew Rest for TAP Portugal</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-02 Glass Screens of Displays and Monitors The requirements for environmental protection and the associated certified noise and/or emissions level...</td>
<td>Marshall ADG Modification Sheet No. MA2094 Issue 1 Must be installed in conjunction with STC EASA 30140007: Installation of new electrical Floor Path Marking System</td>
<td>21/07/2014</td>
<td>Active</td>
<td>10249937</td>
<td></td>
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<tr>
<td>RUIAG AG</td>
<td>10049974</td>
<td>AS350D, AS350BA, AS350DA, AS350B1, AS350BB, AS350B2, AS350BB, AS350BB</td>
<td></td>
<td>Single or dual installation of Garmin GTN 650 GPS/NAV/COMRev 1. Transfer of STC to a new legal entity that is holder of new DOA (EASA 21 J 7896).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS27 Annex 3 paragraphs: 27.1, 27.29, 27.251, 27.301, 27.303, 27.305, 27.307, 27.309, 27.321, 27.337, 27.341. Rev. 6, 27.581, 27.601, 27.603, 27.605, 27.607, 27.639, 27.610, 27.611, 27.613...</td>
<td>Flight Manual Supplement, doc. ref. FMS-690102 Rev. A. alt. 17/07/2014 Master Data List doc. ref. MDL-600147, Rev. A. alt. 17/07/2014 Certification Compliance Sheet doc. ref. CCS-800147, Rev. A. alt. 17/07/2014 Instructions for Continued Airworthiness, doc. ref. ICA-800147, Rev. Addl. 04/07/2014 Hor...</td>
<td>- GPS (GPS use limited to VFR-GPS approaches prohibited-Obstacle and Fuel data only for information)Prior to installation of this design change it must be determined that the new design change and any other previously installed design change and or repair will introduce no adverse...</td>
<td>18/02/2021</td>
<td>Active</td>
<td>10249974</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>STC Holder</td>
<td>FALCON 2000, FALCON 2000EX</td>
<td>Installation of a third Rockwell Collins AHC-300 Altitude Heading Reference System (AHRS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data Package 366-00-0000 revision E, dated August 2011. Master Drawing List 366-00-0001 revision E, dated 03 August 2011.</td>
<td>Applicable to F2000 and F2000EX models with Rockwell Collins Proline 21/Avionics System installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse...</td>
<td>23/07/2014</td>
<td>Active</td>
<td>10409602</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>GV-SP</td>
<td>Installation of an EMS AMT-700G SATCOM Antenna Subsystem.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Instructions for Continued Airworthiness Model GV: CE411030417 rev B dated January 4, 2013 for EMS AviationAMT-700G-01SAT COM antenna subsystem. Models GV, GV-SP: CES52030401 revA dated November 16, 2011 for EMS Aviation AMT-700G-04 SATCOM antennasubsystem or later revisions of the above listed docs.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>24/07/2014</td>
<td>Active</td>
<td>10409606</td>
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<tr>
<td>STZ-AFL</td>
<td>LS 6-B</td>
<td>Retrofit of the Sailplane LS6-b with Winglets</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Description of the modification &quot;STC LS6-b Winglet Retrofit&quot;List of certification documents dated 23.07.2014. Supplement to the Flight Manual of the sailplane LS6-b with retrofitwinglets &quot;LS6-b&quot;-Ergänzung des Segelflugzeug LS6-b mit nachträglichem Winglets &quot;LS6-b&quot;-Drawing...</td>
<td>Reduction of the mass of the non lifting parts by 10 kg compared to the original LS6-b. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>28/07/2014</td>
<td>Active</td>
<td>10500001</td>
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<tr>
<td>MICRO AERODYNAMICS, INC.</td>
<td>402C, 414A</td>
<td>Install of vortex generators on the wing and vertical tail surfaces and air intake in maximum take-off weight, maximum zero fuel weight, minimum ramp weight in accordance with FAA STC SA4934SW as long as FAA STC SA7463SW-D is not yet EASA approved. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>This STC does not cover Cessna 414A equipped with RAM engine and winglet installations per FAA STC SA7463SW-D and SA4934SW as long as this STC FAASTC SA7463SW-D is not yet EASA approved. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>Drawing List DL-402C, Rev. B, dated July 24, 1992Installation Instructions I-402C, Rev. B, dated July 24, 1992Flight Manual Supplement (option 2 only)CG42C-AMFS-1, Rev. B for the 402C, dated August 19, 1992AOA414A-AMFS-1, Rev. B for the 414A, dated August 19, 199290660-00-0014 initial revision, dated 14 April 2011 or later revisions of the above listed...</td>
<td>25/07/2014</td>
<td>Active</td>
<td>10500010</td>
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<tr>
<td>DAO AVIATION A/ S</td>
<td>REV. 1</td>
<td>Installation of Garmin G900PFD/MFD, GPS 500W TAWS, and GTX330D withADS-B out.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:CS-23 Amndt. 3 elected to comply with for areas affected by the change. The requirements for environmental protection and the associated certificated noise and/or emissions lev...</td>
<td>Installation in accordance with DAO Aviation A/ S documentsDAO-DO-0524- MDL, at revision 00, dated July 11, 2014.Aircraft Flight Manual Supplements, DAO Aviation A/S documentsDAO-DO-0524-AMFS1, DAO-DO-0524-AMFS2, and DAO- DO-0524-AMFS3, at revisionH, dated July 11, 2014 or later revisions of the above...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>28/07/2014</td>
<td>Active</td>
<td>10500014</td>
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<td>EASA STC no.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10050018</td>
<td>S-76 A, B &amp; C SERIES</td>
<td>Introduction of Baggage/Cargo to the Cabin Rear Bench Seat Location</td>
<td>Flight Manual IRL Fm Supplement No. 23: &quot;Baggage/Cargo carriage within the Cabin Bench Seat Area&quot; or later revisions of the above listed documents approved by EASA Design Definition.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadworthiness effect upon the airworthiness of the product.</td>
<td>25/07/2014</td>
<td>Active</td>
<td>10050018</td>
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<tr>
<td>10050019</td>
<td>MBB-BK 117 C-2</td>
<td>Installation of GNC 255 and GTN 650 Certification Program STC N 0015020453-001</td>
<td>RFM MBB BK117 C-2 Supplement FMS 10- CRF-C025 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadworthiness effect upon the airworthiness of the product.</td>
<td>25/07/2014</td>
<td>Active</td>
<td>10050019</td>
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<td>10050022</td>
<td>TB 10</td>
<td>Installation of exhaust according to Chabord modification STC- TB10-13-01Rev 3 - Fitting bracket improvement - Enhanced silent blocks - Alternative and corrosion surface treatment - Enhanced muffler collar - Heater improvement (2 in 1)Rev. 4 comp.</td>
<td>Total Technical Dossier ref. STC-TB10-13-01 issue dated 03 November 2021.</td>
<td>This STC is applicable to TB10 aeroplanes equipped with Lycoming O-360-A1A2 engine and with the following propellers: - Hartzell HC-220K-188F/768A-2, - Chevallier B3036C424A7SA-0, - Hoffmann HO-V123C-01(C) or later revisions of the above listed document(s) approved accepted/under the EASA system.</td>
<td>26/11/2021</td>
<td>Active</td>
<td>10050022</td>
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<tr>
<td>10050033</td>
<td>MD900</td>
<td>Lowered Patient Loading System (LPLS)</td>
<td>Flight Manual IRL FMS/707-002 Iss. A or later revisions of the above listed documents approved by EASA Design Definition: (Cert Plan) CP-707-001 Iss. A Compliance Statement document: (CLL) CCL/707-003 Iss. A. MPI/707-004 Iss. A</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadworthiness effect upon the airworthiness of the product.</td>
<td>28/07/2014</td>
<td>Active</td>
<td>10050033</td>
<td></td>
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<tr>
<td>10050044</td>
<td>747-400</td>
<td>RAM cabin reconfiguration and irs “RAVE” IFE installation/Cabin rearrangement and new IFE system installation</td>
<td>CP-B747-12-007 rev 2 (15 Jul 2014) - Certification Program/CLL-B747-12-007 rev 2 (15 Jul 2014) - Compliance Check List/AFMS-B747-12-007 rev 0 (15 Nov 2013) - AIRM Supplement/OPC-B747-12-007 rev 2 (15 Jul 2014) - N°0640 Compliance Checklist - N°0640</td>
<td>- IFE system SCUs (System Control Unit) cellular modem function must be deactivated. - The Safran STC EASA.A.01096 must be installed per drawing 33120002. Initial issue (installation instruction MOS-10 rev F). - The Safran STC EASA.A.01098 charging instructions per AIRM Supplement/AFMS-B747-12-007.</td>
<td>28/07/2014</td>
<td>Active</td>
<td>10050044</td>
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<td>10050079</td>
<td>A321-231</td>
<td>Cabin modification with Zhi Rave IFE system.</td>
<td>Master Data List 4616-01-A-2301-112 REV. 2: Use of the Cellular Modem function of the Zhi “RAVE” IFE/validated.Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadworthiness effect upon the airworthiness of the product.</td>
<td>08/06/2016</td>
<td>Active</td>
<td>10050079</td>
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<td>STC Holder</td>
<td>STC Ref.</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>RHEINLAND AIR SERVICE GMBH</td>
<td>10050088</td>
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<td>CESSNA 208</td>
<td>Installation of single Garmin G600</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 323 ( &gt;7,300 lb)</td>
<td>None.</td>
<td>31/07/2014</td>
<td>Active</td>
<td>10050000</td>
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<tr>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td>10050089</td>
<td></td>
<td>EA 300/1</td>
<td>Introduction of Wing Tip Cameras on Extra EA300/1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>31/07/2014</td>
<td>Active</td>
<td>10050008</td>
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<tr>
<td>AIRBUS S.A.S.</td>
<td>10050094</td>
<td>REV. 1</td>
<td>A320-232</td>
<td>MOD CJ1300 ‘MASTER MOD CJ – AL SAFAR’ EASA STC update for A320 MSN 5255, including Weight and Balance clarification for delivery to OATAR Airways.- Correction on ALS (former ACO) limitations of Compliance IMS CS1300 (to EASA Certification Basis Ref A-2301-008-1): Type correction in Safety equipment.Etc.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>MASTER MINIMUM EQUIPMENT LIST - MMEL-S MOD CJ1300 AUG 01/19/E08 AND BALANCE MANUAL SUPPLEMENT 2</td>
<td>Prior to installation of this design change/repair it must be determined that the interrelationship between this design change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/06/2019</td>
<td>Active</td>
<td>10050094</td>
<td></td>
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<tr>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td>10050103</td>
<td></td>
<td>DHC-8-401, DHC-8-402</td>
<td>Dash 8 Q400 SkyCam Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25, Amendment 4, see Appendix The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged. The Certification Basis is applicable to this certificate/approval.</td>
<td>Installation in accordance with Bournemouth Aviation (Consultants) Ltd/Master Data List BAC-917 MOL02, Issue 2. Maintenance in accordance with Bournemouth Aviation (Consultants) Ltd/Instructions for Continued Airworthiness BAC-917 CAQ02, Issue 2. Maintenance in accordance with Bournemouth Aviation (Consultants).</td>
<td>Provisions installed under Minor Change BAC-917 MOD01 (or staged Minor Changes BAC-917 MOD03, BAC-917 MOD004 and BAC-917 MOD005) are required. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change must be determined.</td>
<td>01/06/2014</td>
<td>Active</td>
<td>10050102</td>
<td></td>
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<tr>
<td>JET AVIATION AG</td>
<td>10050143</td>
<td></td>
<td>MYSTERE FALCON 20- CDS5/ES/FS</td>
<td>TCAS II (ACAS II) UPGRADE TO Version 7.1 upgrade of existing honeywell CAS576 Traffic Alert and Collision Avoidance Systems incorporate 7.1 log software,Change consists of Honeywell TPU-47A TCAS 7.1 processor P/N N66-01168-121 upgrade to complement by a TPU-47B TCAS 7.1 processor P/NM6-01168-222...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List JBSL-M-00057MC-01 Issue 1/Instructions for Continued Airworthiness JBSL-M-0007CA-01 Issue 1/ Airplane Flight Manual Supplement JBSL-M-0007AFM-01 Issue 1/ later revisions of the above listed documents approved by EASA.</td>
<td>As a pre-require, airplane must be fitted with existing ACAS II compliant TCAS-Honeywell TPX50a P/N 065-01168-121, and EHS and/or ADS-B compliant SSR ATC transponders, and altitude data not provided through Gray/ Gilham coding. Prior to installation of this design change it must be determined that...</td>
<td>06/08/2014</td>
<td>Active</td>
<td>10050143</td>
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<tr>
<td>TAMAC AEROSPACE SWITZERLAND AG</td>
<td>10050151</td>
<td></td>
<td>737-800</td>
<td>WLAN Activation/This Modification activates the WiFi access point built into the aircraft within the 2.4 GHz and 5 GHz frequency bands. In addition a Cockpit Control Panel — WLAN ON/OFF.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, Doc. No. MOL-AR07-4430, Rev. B/AFMS, Doc. No. AFMS-AR07-4430 Rev. A / ICA Doc. No. ICA-AR07-4430, Rev. A/other revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/08/2014</td>
<td>Active</td>
<td>10050151</td>
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<td>STC Holder</td>
<td>REV.</td>
<td>Foreign/NAA STC(s)</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST0969NY</td>
<td>FALCON 2000, FALCON 2000EX</td>
<td>Validation of FAA STC Installation of EMTEO LED Anti-Collision / Navigation Light.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List ref. F2XW-D0005-005 issue A, dated 21/06/2013.</td>
<td>Applicable to F2000 model with Dassault Aviation Service Bulletin No. 34</td>
<td>02/12/2014</td>
<td>Active</td>
<td>10050156</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST0169CH-D</td>
<td>MF 50</td>
<td>Validation of FAA STC Installation of Wielkopolski Single-Function LED Navigation Tail Light.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List ref. 10226B66 issue C, dated 07/12/2002 for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decs.</td>
<td>NONE</td>
<td>07/08/2014</td>
<td>Active</td>
<td>10050159</td>
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<tr>
<td>SCHERDEL ENGINEERING LTD</td>
<td>REV. 1</td>
<td>2DG 500MB - 1DG 800A / -1DG 800B</td>
<td>Alternate Method of Compliance for TM 800/42 and TM500/06</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Alternate Method of Compliance TM 800-42, 500-06, Doc.No TN 5, V3 - additional pages to Flight Manual, chapter for later revisions of the above listed documents approved by EASA.-Certification Document, Rev. 1: Alternate Method of Compliance for TM 800/42, 500/06, CD, V1</td>
<td>Only for DG 800A, DG 800B (EASA.A.067)DG 500MB (EASA.A.233)</td>
<td>07/08/2014</td>
<td>Active</td>
<td>10050163</td>
<td></td>
<td></td>
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<tr>
<td>LAMITZ PRENA FOLIEN FACTORY GmbH</td>
<td>REV. 5</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Application of ORATEX covering and coating system for sailplanes</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:the following paragraph(s) at a later amendment: CS 22.3, CS 22.4, CS 22.45, CS 22.83, CS 22.143, CS22.152, CS 22.201, CS 22.202, CS 22.203 CS 22.207, CS 22.251, CS 22.301.</td>
<td>- Application of ORATEX covering &amp; coating system, Doc. No. ADX-C-51-001-AMM Edition 8.0, dated 6th March 2020.-Implementation Procedures of EU/US Bilateral Agreement.</td>
<td>Prior to installation of this modification, it must be determined that the installation of this modification to aircraft and any other previously installed modification does not interfere with the airworthiness of the product.</td>
<td>02/06/2020</td>
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<td>10050167</td>
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<td>10050194</td>
<td>AVIONIK</td>
<td>REV. 1</td>
<td></td>
<td>340</td>
<td>Installation of Garmin G3X, Garmin G3X, L-3 Communications NY-163, Comart CI 690-1, LH and RH Sensor Systems S87-2002</td>
<td>Master Document List A5R-2014-016-MDL-01-01 (Rev. 01) dated 29th June 2017</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/02/2017</td>
<td>Active</td>
<td>10050194</td>
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<tr>
<td>10050196</td>
<td>AVIONIK</td>
<td>REV. 1</td>
<td></td>
<td>340</td>
<td>Mast with relay antenna installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
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<td>10050203</td>
<td>ASI INNOVATION</td>
<td>REV. 1</td>
<td></td>
<td>340</td>
<td>Installation of Cabin Door Step System</td>
<td>Asis-APM/012-06 (aircraft flight manual supplement issued 08/2014) or later revisions of the above listed documents approved by EASA.</td>
<td>Other: maximum take-off weight reduced by 800 lbs (5800 Ib) noise certified 80 dB (maximum noise authorized).</td>
<td>13/06/2014</td>
<td>Active</td>
<td>10050203</td>
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<tr>
<td>10050209</td>
<td>DOA21 LIMITED</td>
<td>REV. 1</td>
<td></td>
<td>340</td>
<td>Cabin refurbishment including LED Cabin lighting installation.</td>
<td>STC Twenty One Flight Manual Supplement No. S21-FMS-011 Issue for later revisions of the above listed documents approved by EASA</td>
<td>Restricted to MSN 5538.</td>
<td>27/09/2014</td>
<td>Active</td>
<td>10050209</td>
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<tr>
<td>10050212</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING INC</td>
<td>REV. 1</td>
<td></td>
<td>340</td>
<td>Introduction of Corporate VIP Interior/Rev. 1: Transfer of STC holder role to DOA21 Ltd.</td>
<td>STC Twenty One Flight Manual Supplement No. S21-FMS-011 Issue for later revisions of the above listed documents approved by EASA</td>
<td>Restricted to MSN 5538.</td>
<td>24/10/2023</td>
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<td>10050216</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING INC RPRORATION AG (AIEC)</td>
<td>REV. 1</td>
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<td>340</td>
<td>Dual MD302 SAM Standby Attitude Module</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10050217</td>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD</td>
<td>REV. 1</td>
<td>TCGA STC O-LSA11-222/D</td>
<td>BD-700-1A10</td>
<td>I-KU System Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Innotech Aviation Division of IMP Group Ltd. Flight Manual Supplement FM11-006, Revision A, Transport Canada approved 9th December 2011 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement.</td>
<td>Restricted to ASN0398</td>
<td>20/10/2014</td>
<td>Active</td>
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<tr>
<td>10050223</td>
<td>AEROSPACe DESIGN AND CERTIFICATION</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING 727-252F (RE)</td>
<td>Boeing 727 Oil Dispersant Equipment Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Paragraph 4 Airworthiness Limitations, Reference JN01/002 ADC 54(2) Issue 2 or later revisions of the above listed documents approved by the FS.</td>
<td>AERADAC Aircraft Flight Manual Supplement FM/ JN/491-01/050 Issue E, Reference JN 491/01/002, ADC 29(1) Issue E AERADAC Instructions for Continued Airworthiness, Paragraph 4 Airworthiness Limitations, Reference JN01/002 ADC 54(2) Issue 2 or later revisions of the above listed documents approved by EASA.</td>
<td>Aircraft Serial Number 22938</td>
<td>14/08/2014</td>
<td>Active</td>
<td>10050223</td>
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<td>10050229</td>
<td>CRANE AEROSPACe &amp; ELECTRONICS</td>
<td>REV. 1</td>
<td>S550 CITATION SXI CIVILATION VII 680 SXI CIVILATION VII 690 XLS EXCEL AND XLSI 680 XLSI 690 EXCED 690 ENCORE 690 ENCORE + 690 CITATION II AND BRAVO 695 CITATION AND CITATION I</td>
<td>Tire Pressure Sensor on Cessna Citation Model Installation of a Crane Aerospace &amp; Electronics Tire Sensor Assembly</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Crane Aerospace, Installation Manual, I83-201-01, rev. 1 dated 11/2013 Crane Aerospace, Master Document List, CCL85-201-01, rev. C, dated 02/2016 Including the Instructions for Continuous Airworthiness (ICA) or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>03/02/2015</td>
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<td>10050229</td>
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<td>10050244</td>
<td>FR AVIATION LTD/D/BA COBHAM AVIATION SERVICES UK</td>
<td>CAA UK RESPONSIBILITY</td>
<td>FAN JET FALCON E FAN JET FALCON D FAN JET FALCON</td>
<td>Introduction of an Observers Seat</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Crane Airworthiness and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Cobham Aviation Services Aircraft Flight Manual Supplement No. FR/AA/FM/ 0126 dated 08/08/14 or later revisions of the above listed documents approved by EASA.</td>
<td>Cobham Aviation Services Aircraft Flight Manual Supplement No. FR/AA/FM/ 0126 dated 08/08/14 or later revisions of the above listed documents approved by EASA.</td>
<td>Cobham Aviation Services Minor Modification No. DC-0089 is Pre-Required for this installation</td>
<td>19/08/2014</td>
<td>Active</td>
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<td>10050247</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>A320-232</td>
<td>Activation of Wireless IFE System Board Connect with Gate-to-Gate functionality.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental Type Definition Document STD025-15/5M Issue 1 Rev. 0, dated 24/09/2015- Compliance Record List CRL25-15/5M Issue 1 Rev. 0, dated 24/09/2015- Airplane Flight Manual Supplement A-00-15/46-4FM Issue 1 Rev. 0, dated 16/09/2015- Instructions for Continued Airworthiness A-00-15/46-ICA Issue...</td>
<td>None.</td>
<td>29/09/2015</td>
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<td>EASA STC No.</td>
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<td>Certification basis</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10050248</td>
<td>ARMSTRONG AEROSPACE Inc.</td>
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<td>A330-200</td>
<td>Installation of a Lufthansa Systems BoardConnect IFE System. The BoardConnect In Flight Entertainment (IFE) system delivers access to information and entertainment for passengers via an onboard Wi-Fi network, which utilizes passenger operated personal electronic devices (PED).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Refer to the appended document with the certification number 10050248, dated 16 July 2014. The requirements for environmental protection and the associated certified noise and/or emissions level...</td>
<td>Master Data List: MDL DC 104-9341-00, rev. 8, FAA: DC 113-9341-00 rev. AAFM: DC 303-9341-00 rev. 12 FAA Letter (No reference) dated 13 August 2014 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/94/EC (or subsequent revisions of this decision).</td>
<td>20/08/2014</td>
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<td>10050266</td>
<td>HELI-ONE CANADA LLC</td>
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<td>TCCA STC SH12-25 AS 332 L, AS 332 L1</td>
<td>Installation of Enhanced Ground Proximity Warning system Honeywell MarkXII EGPWS (This STC is the validation of TCCA STC SH12-25 issued on September 12, 2012.)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Validation of TCCA STC SH12-25 issued September 12, 2012. (or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canada Bilateral Agreement)...</td>
<td>None</td>
<td>22/08/2014</td>
<td>Active</td>
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<td>10050267</td>
<td>HELI-ONE</td>
<td>REV.1</td>
<td></td>
<td>TCCA STC SH12-25 ISSUE 2 AS 332 L, AS 332 L1</td>
<td>Installation of IFE (In Flight Entertainment) system to be installed as per EASA STC 10058357. Prior to installation of this design change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Configuration 1 (Single FMS installation).</td>
<td>Configuration 2 of the IFE system to be installed as per EASA STC 10058357. Prior to installation of this design change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/08/2016</td>
<td>Active</td>
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<tr>
<td>10050272</td>
<td>PW ENGINEERING GmbH</td>
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<td></td>
<td>A330-200</td>
<td>Installation of a Wireless System in the A330-200 VP786. This STC installs a new Wireless System in replacement of an existing one (same location through the aircraft). The system includes an antenna unit, wireless access points, control panel in cabin switch box...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement, ref A332A-AMF-A dated 17 July 2014...</td>
<td>This approval is limited to MSN 487. Prior to installation of this design change, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/08/2014</td>
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<td>10050279</td>
<td>ARMSTRONG AEROSPACE Inc.</td>
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<td>FAA STC ST03333CH</td>
<td>Installation of Cobham SATCOM System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Master Data List, Armstrong Aerospace document No. DC104-9341-00, Revision C.&lt;...&gt; dated 1 May 2014: Instructions for Continued Airworthiness: Armstrong Aerospace document No. DC113-9341-00, Revision B, dated 30 April 2014: Arm...</td>
<td>Modification in accordance with this STC requires prior or concurrent installation of Delta G Datas &amp; Design, Inc. STC ST11265C, validated by EASA...</td>
<td>25/08/2014</td>
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<td>STC No.</td>
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<td>10050321</td>
<td>AVIONIK STRAUBINGENTWICKLUNGS GmbH</td>
<td>340A 340</td>
<td></td>
<td>Installation of Garmin G600 System with Synthetic Vision (SVT)</td>
<td>The Certification Basis for the original product as amended by: A suitable Garmin GPS System must have already been installed.</td>
<td>Flight Manual Supplement G600 ASR-2014-038-FMS-05-00, Rev. 00.</td>
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<td>27/06/2014</td>
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<td>10050325</td>
<td>STC TWOY ONE LIMITED</td>
<td>BD-700-1A10</td>
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<td>Refreshment of VIP Interior</td>
<td>The Certification Basis for the original product as amended by: Restricted to ASB 9038 EASA STC No. 100/1640 Rev 4 is a pre-requisite for this modification.</td>
<td>STC Twenty One Flight Manual Supplement No. S21-FMS-0112 Issue 1 or later revisions of the above listed documents approved by EASA STC Twenty One Master Document List No. S21-MDL-1188 Issue 1</td>
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<td>27/08/2014</td>
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<td>10050326</td>
<td>SWIS CERTIFICATION SERVICES ltd.</td>
<td>B777-200ER</td>
<td></td>
<td>ANZ B777-200 Skycoach Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>SWIS Certification Services Ltd Master Configuration Scheme No.3101-MCS-01 Issue 1</td>
<td>None.</td>
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<td>27/08/2014</td>
<td>Active</td>
<td>10050326</td>
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<tr>
<td>10050344</td>
<td>ARMSTRONG AEROSPACE Inc.</td>
<td>A321-212 / -213 / -231 / -232 A321-111 / -112 / -131 / -132 A320-231 / -232 / -233 A320-211 / -213 / -214 A319-115 / -131 / -132 / -133 A318-111 / -112 / -113 / -114</td>
<td>Certification of Compatibility with Cellular and Wireless Data Technologies.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA STC ST03247CH, issued August 20, 2013 Master Data List DC10A-0214-00, Revision 1, dated August 9, 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequently revised decisions) and/or the Technical Implementation Proc.</td>
<td>FAA STCNo. ST03247CH or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions off this decision) and/or the Technical Implementation Proc.</td>
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<td>10050359</td>
<td>AERO-DIENST GMBH &amp; C. KG</td>
<td>CL-600-2B16</td>
<td>Satcom Antenna AMT-1800 IGA Installation (Honeywell)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA STC ST03247CH, issued August 20, 2013 Master Data List DC10A-0214-00, Revision 1, dated August 9, 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequently revised decisions) and/or the Technical Implementation Proc.</td>
<td>None.</td>
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<td>01/09/2014</td>
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<td>10050359</td>
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<td>10050377</td>
<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>REV. 1 FAA STC ST0266WY</td>
<td>MF50/MF900/F900EX</td>
<td>Validation of FAA STC Installation of EMTEQ LED Anti-Collision Navigation Light. This Rev. 1 was issued to correct the STC approval holder name.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List ref. F900X-00007-005 Rev. B dated 20 November 2013, Airworthiness Limitation Section as per Maintenance Manual Supplement ref. F900X-00007-155 Rev. C, dated 24 October 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF</td>
<td>Applicable to MF50 model with Dassault Aviation Service Bulletin No. 48 embedded, Applicable to F900EX model with Dassault Aviation Service Bulletin No. 385 embedded</td>
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**List of EASA Supplemental Type Certificates**

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<td>Installation of Garmin G600 System with Synthetic Vision (SVT)</td>
<td>The Certification Basis for the original product as amended by: A suitable Garmin GPS System must have already been installed.</td>
<td>Flight Manual Supplement G600 ASR-2014-038-FMS-05-00, Rev. 00.</td>
<td></td>
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<td>10050325</td>
<td>STC TWOY ONE LIMITED</td>
<td>BD-700-1A10</td>
<td></td>
<td>Refreshment of VIP Interior</td>
<td>The Certification Basis for the original product as amended by: Restricted to ASB 9038 EASA STC No. 100/1640 Rev 4 is a pre-requisite for this modification.</td>
<td>STC Twenty One Flight Manual Supplement No. S21-FMS-0112 Issue 1 or later revisions of the above listed documents approved by EASA STC Twenty One Master Document List No. S21-MDL-1188 Issue 1</td>
<td></td>
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<td>27/08/2014</td>
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<td>10050326</td>
<td>SWIS CERTIFICATION SERVICES ltd.</td>
<td>B777-200ER</td>
<td></td>
<td>ANZ B777-200 Skycoach Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>SWIS Certification Services Ltd Master Configuration Scheme No.3101-MCS-01 Issue 1</td>
<td>None.</td>
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<td>27/08/2014</td>
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<tr>
<td>10050344</td>
<td>ARMSTRONG AEROSPACE Inc.</td>
<td>A321-212 / -213 / -231 / -232 A321-111 / -112 / -131 / -132 A320-231 / -232 / -233 A320-211 / -213 / -214 A319-115 / -131 / -132 / -133 A318-111 / -112 / -113 / -114</td>
<td>Certification of Compatibility with Cellular and Wireless Data Technologies.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA STC ST03247CH, issued August 20, 2013 Master Data List DC10A-0214-00, Revision 1, dated August 9, 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequently revised decisions) and/or the Technical Implementation Proc.</td>
<td>FAA STCNo. ST03247CH or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequently revised decisions) and/or the Technical Implementation Proc.</td>
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<td>10050359</td>
<td>AERO-DIENST GMBH &amp; C. KG</td>
<td>CL-600-2B16</td>
<td>Satcom Antenna AMT-1800 IGA Installation (Honeywell)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA STC ST03247CH, issued August 20, 2013 Master Data List DC10A-0214-00, Revision 1, dated August 9, 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequently revised decisions) and/or the Technical Implementation Proc.</td>
<td>None.</td>
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<td>01/09/2014</td>
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<td>10050377</td>
<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>REV. 1 FAA STC ST0266WY</td>
<td>MF50/MF900/F900EX</td>
<td>Validation of FAA STC Installation of EMTEQ LED Anti-Collision Navigation Light. This Rev. 1 was issued to correct the STC approval holder name.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List ref. F900X-00007-005 Rev. B dated 20 November 2013, Airworthiness Limitation Section as per Maintenance Manual Supplement ref. F900X-00007-155 Rev. C, dated 24 October 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF</td>
<td>Applicable to MF50 model with Dassault Aviation Service Bulletin No. 48 embedded, Applicable to F900EX model with Dassault Aviation Service Bulletin No. 385 embedded</td>
<td>05/12/2014</td>
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The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

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<th>EASA STC no.</th>
<th>STC Holder</th>
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<th>Foreign/NAAC/STC ref</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<td>10050434</td>
<td>H4 AEROSPACE (UK) Ltd</td>
<td>REV 1</td>
<td>FAA ST03185NY</td>
<td>A340-642, A340-643, A340-541, A340-542, A340-311, A340-312, A340-313, A340-211, A340-212, A340-213</td>
<td>Honeywell/TCAS II system upgrade to change 7.1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Modification Summary Sheet HA41742, dated 19/09/2017; - Instructons for Continued Airworthiness H4ADR854 Issue 3, dated 19/09/2017; - Airplane Flight Manual Supplement H4AFMS039 Rev 2, dated 19/09/2017</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noiseless effect upon the airworthiness of the product.</td>
<td>02/10/2017</td>
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<td>10050443</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV 1</td>
<td>FAA ST03185NY</td>
<td>A321-213, A321-232, A321-131, A321-211, A321-212, A321-232, A321-111, A321-112, A320-214, A320-231, A320-232, A320-111, A320-211, A320-212, A319-131, A319-132, A318-133, A318-139, A318-114, A318-115, A318-132, A318-111, A318-112, A318-112, A318-121</td>
<td>Installation of UTC Electronic Flight Bag Display Mount for Captain and first officer</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, ref: 0164-0885-2367</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce noiseless effect upon the airworthiness of the product.</td>
<td>29/04/2019</td>
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<td>10050453</td>
<td>SPECTRUM AEROMED, LLC/ via SPECTRUM AEROMED</td>
<td>REV 2</td>
<td>FAA STC SA03345CH</td>
<td>P 180 AVANTI P 180 AVANTI II</td>
<td>Installation of Air Ambulance Conversion into P 180 Avanti or P 180Avanti II in accordance with Spectrum Aeromed Master Drawing List054DL, Revision 1, dated 29 May 2014</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Spectrum Aeromed Master Drawing List, Doc. No. 6054FMS, Revision 1, dated 29-May-2014; FAA approved Aircraft Flight Manual Supplement, Doc. No. 6054FM, Original Issue, dated 08-Jan-2014; FAA accepted Instructions for Continued Airworthiness, Doc. No. 6054FM, Revision 1, dated 07-Feb-2013 or later revisions.</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce noiseless effect upon the airworthiness of the product.</td>
<td>24/04/2019</td>
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<td>DART AEROSPACE Ltd.</td>
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<td>TCCA SH13-43</td>
<td>R 66</td>
<td>DO06-1001 Bearpaw Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Instructions IN-DO06-1001 Revision B, dated November 13, 2013 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canada Bilateral Agreement Instructions for Continued Airworthiness ICA-DO06-1001 revision 0, dated 24/04/2014.</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce noiseless effect upon the airworthiness of the product.</td>
<td>12/09/2014</td>
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<td>10050473</td>
<td>VECTOR AEROSPAC EF HELICOPTER SERVICES INC.</td>
<td>FA- SR02399AK</td>
<td>A53 322 C, A53 322 L, A53 322 L1</td>
<td>Garme GNSS530W Installation in accordance with Vector Aerospace/ Helicopter Services USA Inc. Master Drawing List HP 332-34-60 -102Rev.A. This STC is the validation of FAA STC SR02399AK.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI F-01 “HIRF Protection.” The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>15/09/2014</td>
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<td>10050483</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>1900D</td>
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<td>TCAS antenna relocation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Installation and maintenance according to the documentation/report in the Master Data List AE14151MDL Rev. 0 dated 10/09/2014 or later revisions of the above listed documents approved by EASA.</td>
<td>15/09/2014</td>
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<td>10050501</td>
<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>AVRO 146 SERIES RJ85 AVRO 146 SERIES RJ70 AVRO 146 SERIES RJ100</td>
<td>LPV Approach</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI H-01 Electronic Wiring Interconnection System (EWIS). The requirements for environmental protection and the associated certified noise and/or emissions remain applicable.</td>
<td>Suitable with the operation to LPV Approach capability. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>16/09/2014</td>
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<td>10050506</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 1</td>
<td>2-FR182 1-R182</td>
<td>C182 MOGAS Operations/ Installation of a modification kit to allow the operation with MOGAS as a engine. The aircrafts are to be operated i.e. the Flight Manual Supplement as listed below Rev. 1 Transfer to ACC Columbia Jet Service GmbH</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI H-01 Electronic Wiring Interconnection System (EWIS). The requirements for environmental protection and the associated certified noise and/or emissions remain applicable.</td>
<td>Master Document List MODDOC_2033441_rev0_20140715 (Flight Manual Supplement) SPOH_2033441_rev20140715 (Flight Manual Supplement)</td>
<td>According to limitation section of AFM Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product if the ho...</td>
<td>31/01/2022</td>
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<td>POLYTECH AERO SUPPORT INC.</td>
<td>REV 1</td>
<td>TCA SA14-33, ISSUE 1</td>
<td>A320-341, A320-342, A320-343, A320-301, A330-321, A330-332, A330-202, A330-343</td>
<td>A330 Improved Air Filter Element installation/This modification is to substitute to the originally installed air-conditioning filter elements by improved air filter elements, manufactured by IDP inc., without modifying the system original design and associated maintenance procedures.</td>
<td>The Certification Basis for the original product as amended and/or other previously installed design change and/or repair will introduce maneuver factor upon the airworthiness of the product.</td>
<td>Pre-installation of this design change must be determined by the technical interchangeability between this design change and any other previously installed design change and/or repair.</td>
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<td>AIRWORTHINESS TECHNOLOGY S.L.</td>
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<td>SA227-8C</td>
<td>SA227-AC</td>
<td>Replacement of the original Aircraft Ni-Cd batteries to install ConcordeRG-390E sealed lead-acid batteries.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the technical interchangeability between this design change and any other previously installed design change and/or repair will introduce maneuver factor upon the airworthiness of the product.</td>
<td>19/09/2014</td>
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<td>LUFTHANSA TECHNIK AG</td>
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<td>737-300, 737-600</td>
<td>Upgrade of the ACARS CMU software – Installation of ACA-Ready software configuration.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the technical interchangeability between this design change and any other previously installed design change and/or repair will introduce maneuver factor upon the airworthiness of the product.</td>
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<td>A321-231, A321-232, A321-211, A321-212, A321-213, A321-111, A321-112, A321-131, A320-231, A320-232, A330-233, A320-214, A320-215, A320-216, A320-211, A320-212, A319-122, A319-133, A319-14, A319-114, A319-115, A319-116, A319-111, A319-112, A319-113</td>
<td>Tablet Interface Module (TIM) Installation</td>
<td>The Certification Basis for the original product as amended and/or other previously installed design change and/or repair will introduce maneuver factor upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined that the technical interchangeability between this design change and any other previously installed design change and/or repair will introduce maneuver factor upon the airworthiness of the product.</td>
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<td>TENENCIA Ltd.</td>
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<td>CAA UK RESPONSIBIL ITY</td>
<td>A310-304</td>
<td>Modification TEN 701 - Airbus A310-304 GPS activation &amp; Thermo Modification activates a Dual OMA-5024 GPS System - previously installed and deactivated pending Operational Requirements for PRM/Non Precision Approach (NPA) - in accordance with Tenencia Ltd report TEN 701 Issue 4, dated 21 August 2...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the technical interchangeability between this design change and any other previously installed design change and/or repair will introduce maneuver factor upon the airworthiness of the product.</td>
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<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U.</td>
<td>B. 4212, 4214, 4216</td>
<td>REV. 3</td>
<td>Boeing 737-500</td>
<td>Cargo Bag</td>
<td>INggeration of Replacement Cargo Bag</td>
<td>Boeing 737-500</td>
<td>10/02/2019</td>
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<td>10050582</td>
<td>GLOBAL AVIATION TECHNOLOGIES LLC</td>
<td>FAA ST01743W</td>
<td>REV. 1</td>
<td>DHC-8-314, 315</td>
<td>Cargo Bag</td>
<td>Installation of Cargo Bag in DHC-8</td>
<td>DHC-8-314, 315</td>
<td>10/01/2014</td>
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<td>10050612</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERSERVING WALTER STARZACHER Gesellschaft für Luftfahrttechnik mbH</td>
<td>737-8</td>
<td>REV. 3</td>
<td>737-300</td>
<td>Cargo Bag</td>
<td>Installation of Cargo Bag in Boeing 737-300</td>
<td>737-300</td>
<td>18/12/2018</td>
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<td>BOMBARDIER Inc.</td>
<td>CL600-2B16</td>
<td>REV. 1</td>
<td>Introduction of Replacement Stand by Altimeter STC transfer</td>
<td>Cargo Bag</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate</td>
<td>CL600-2B16</td>
<td>23/06/2015</td>
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<td>10050616</td>
<td>AERO-DENIST GMBH &amp; CO. KG</td>
<td>CL-600-2B16</td>
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<td>Aspire 200 Satellite System installation (Honeywell)</td>
<td>Cargo Bag</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate</td>
<td>10050616</td>
<td>25/09/2014</td>
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<td>AVIONTEC AG</td>
<td>M20K</td>
<td>M20J</td>
<td>B773</td>
<td>STC Holder</td>
<td>System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Flight Manual Supplement No. 3754-011 Rev.1 of 11.06.2014 Modification Instruction No. 3754-011 Rev.0 of 11.06.2014 or later revisions of the above listed documents approved by EASA. Master Document List No. 3754-017 Rev 0 of 11.06.2014 for Continued Airworthiness No. 3754-017 Rev ...</td>
<td>29/06/2014</td>
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<td>10050643</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING 727-212</td>
<td>Installation of a Dual Iridium Antenna</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: FAR 25 at Original Issue 25.1(a), 25.25(a), 25.27(b), 25.29(a)/(b)/(c)/(d)/(e), 25.30(c), 25.305(a)/(b)/(c)/(d), 25.311, 25.321, 25.351/d, 25.365, 25.471, 25.601, 25.605.</td>
<td>Instructions for Continued Airworthiness BAC-928CA01 Issue 3, Boeing 727-212(&lt;-&gt;) Installation of a Dual Iridium Antenna or later revisions of the above listed documents approved by EASA.</td>
<td>Aircraft Serial Number: 21460 only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/06/2014</td>
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<td>10050666</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 2</td>
<td>777-300ER</td>
<td>B773 BEST Cabin reconfiguration [Installation of IFE Equipment Class seats (Oasis); Zodiac; SeaFrance Business class (Cirrus) and Economic class seats (Z301)]; Modification to IFE/E Aerospace Premium Economic class seats. Installation of new partitions and storage as well as modification of existing...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s)- CRI D-02 (Single Side Facing Seats); CRI D-03 (Seats with Inflatable Restraints). The requirements for environmental protection and the associated...</td>
<td>- Change Approval Sheet, Air France document No. CAAS-777-12-007 Revision 4, dated 12 August 2015 - Aircraft Maintenance Manual Supplement, Air France document No AMMS-777-12-007-01, Revision 0, dated 31 July 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>AVIATION DEFENSE SERVICE</td>
<td>406</td>
<td>AVD-MF406-95-ISR1A Issue 01 - Carriage of Airbus Dn't-&gt;&amp;&lt;-&gt;&gt;ISR-406-R1 systems</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>AFM Supplement SAFM- AVD-F406-ISR ed1 dated 24 Sept 2014 - Installation of the above listed documents approved by EASA.</td>
<td>Limitation to serial number 0013.</td>
<td>30/09/2014</td>
<td>Active</td>
<td>10050676</td>
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<tr>
<td>10050677</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>747-8</td>
<td>New Premium Economy Class (B747-8) in the main cabin of the main aircraft the change consists of the removal of 10 per cent of the Economy Class Seats between door 35 and 4 door and the installation of new developed Premium Economy Seatcabin. The new seating will have a greater pitch...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document STDD25-13/37M Issue 1 rev 1. ICA, rev A-00-144-14/AICA, rev 1 rev 0- Compliance Record List CRL25-13/37M Issue 1 rev 2or later revisions of the above listed documents approved by EASA.</td>
<td>This modification must be installed in conjunction with SB AIB88474-25-172 (Introduction of a three part class divider installation) according to the design/ Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/09/2014</td>
<td>Active</td>
<td>10050677</td>
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<td>10050685</td>
<td>ARMSTRONG AEROSPACE</td>
<td>FAA STC</td>
<td>ST0332CH</td>
<td>A320-214</td>
<td>Installation of a Tune Box wireless In-Flight Entertainment (IFE) System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, Armstrong Aerospace document No. DC104-9281-06, Revision D, dated 8 May 2014 - Instructions for Continued Airworthiness, Armstrong Aerospace document No. DC113-9281-06, Revision D, dated 5 May 2014 - Aircraft Flight Manual Supplement, Armstrong Aerospace document No. DC303-9281-00, Revision...</td>
<td>- Modification in accordance with this STC requires prior or concurrent installation of Armstrong Aerospace STC ST0332CH; validated by EASA. Installation of a Cobham S88 SATCOM System. Modification in accordance with this STC requires prior concurrent installation of Armstrong Aerospace STC ST03247...</td>
<td>01/10/2014</td>
<td>Active</td>
<td>10050685</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC Status</td>
<td>Rev. level</td>
<td>Foreign/NAA LDC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10050690</td>
<td>SWCS CERTIFICATION SERVICES Ltd.</td>
<td>Active</td>
<td>1</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>SWS/C/313 - STG AEROSPACE PHOTOLUMINESCENT UNIVERSAL SYMBOLIC EXIT IDENTIFIER SIGNS</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements specifies Condition(s): CRI D-30/09a; CS 25Amdt 14The requirements for environmental protection and the associated certified noise and/or emissions levels of the original...</td>
<td>-Master Configuration Scheme 3134-MCS-01 Iss 3 dd. 30 September 2014 for Continued Airworthiness per Maintenance ManualSupplement 3134- MMS-01 Iss 2 dd. 29 September 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>-THE PHOTOLUMINESCENT EMERGENCY EXIT SIGNS OF THIS STC MAY ONLY BE USED TO REPLACE SIGNS MOUNTED ON ITEMS SUCH AS BULKHEADS, CLASS DIVIDERS ETC IN CABIN OCATIONS AWAY FROM EMERGENCY EXITS. SUCH EMERGENCY EXIT SIGNS ARE NOT REQUIRED BY CERTIFICATION REGULATIONS TO BE ELECTRICALLY ILLUMINATED ALTHOUGH...</td>
<td>01/10/2014</td>
<td>Active</td>
<td>10050690</td>
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<tr>
<td>10050691</td>
<td>ICELANDAIR EHF</td>
<td>Active</td>
<td>10050690</td>
<td>Satellite Communication Systems InstallationRev. 1: Transfer of STC to Passenger Configuration with Avio ZEUS Lie Flat Seats</td>
<td>757-200</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Change/Repair Approval, Icelandair document No. CRA-757-2520-407; Revision A, dated 30 September 2014, ContinuedAirworthiness Instructions, Icelandair document No.CA-S/25-2520-407&lt; (&gt;,&lt;) dated 30 September 2014 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/10/2014</td>
<td>Active</td>
<td>10050691</td>
</tr>
<tr>
<td>10050697</td>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>REV. 1</td>
<td>734</td>
<td>Master Configuration Scheme 3134-MCS-01 Iss 2, dated 30 September 2014 or later revisions of the above listed documents.</td>
<td>BD-700-1A10</td>
<td>Satellite Communication Systems InstallationRev. 1: Transfer of STC to Passenger Configuration with Avio ZEUS Lie Flat Seats</td>
<td>-Master Configuration Scheme 3134-MCS-01 Iss 3 dd. 30 September 2014 for Continued Airworthiness per Maintenance ManualSupplement 3134- MMS-01 Iss 2 dd. 29 September 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Aircraft Serial Number 9098 OnlyPrior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/11/2021</td>
<td>Active</td>
<td>10050697</td>
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<tr>
<td>10050705</td>
<td>SWCS CERTIFICATION SERVICES Ltd.</td>
<td>Active</td>
<td>10050705</td>
<td>Master Configuration Scheme 3134-MCS-01 Iss 2 Skycoach Installation</td>
<td>777-300 ER</td>
<td>China Airlines Boeing 777-300ER Skycoach Installation</td>
<td>-Master Configuration Scheme 3134-MCS-01 Iss 2</td>
<td>- Master Document List No= SWS/C/3134-MCS-01 Iss 2</td>
<td>02/10/2014</td>
<td>Active</td>
<td>10050705</td>
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<tr>
<td>10050732</td>
<td>DART AEROSPACE Ltd.</td>
<td>Active</td>
<td>10050732</td>
<td>Master Configuration Scheme 3134-MCS-01 Iss 2 Skycoach Installation</td>
<td>586-1002 Cargo Mirror Installation</td>
<td>D066-1002 Cargo Mirror Installation</td>
<td>Installation Instructions IAA-I/66-1002 Revision B, dated 13 November 2013 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canalization Agreement - Master Document List No= D066-1002-1, Revision B, 31 July 2014 - Amend...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/10/2014</td>
<td>Active</td>
<td>10050732</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC ref.</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10050735</td>
<td>PAUSTIAN ARTEX GmbH</td>
<td>A321-211</td>
<td></td>
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<td></td>
<td>Business Class Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/authorised noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Report DR-DP07 Rev. 00, dated 04/10/2014</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/10/2014</td>
<td>Active</td>
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<tr>
<td>10050741</td>
<td>DAO AVIATION A/S</td>
<td>B200, B200C</td>
<td></td>
<td></td>
<td></td>
<td>Installation of Dual Garmin GTX-33D transponders for Elementary Surveillance, including ADS-B Out Functionality under DAO modification DAO-DO-0540</td>
<td>As defined in DAO document ref. DAO-DO-0540-CIP-02 dated 2.10.2014. The requirements for environmental protection and the associated/authorised noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with DAO Aviation A/S modification (SAO) as defined in Engineering Change Order DAO-DO-0540-ECO-01 dated 25.09.2014. Operation in accordance with DAO Aviation Flight Manual Supplement (ACMS) DAO-DO-0540-AMS-01 dated 02.10.2014.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/10/2014</td>
<td>Active</td>
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<tr>
<td>10050748</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA SH14-28</td>
<td>R 66</td>
<td></td>
<td>D066-1003 Interior Protector Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/authorised noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Instructions IN-D066-1003, Revision A or later and/or the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EUT Canada Bilateral Agreement. - Master Document List MDL-D066-1003-1, Revision B, dated 31 July 2014. - Aerospace Compliance...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/10/2014</td>
<td>Active</td>
<td>10050748</td>
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<tr>
<td>10050751</td>
<td>TESTORI AERO SUPPLY S.R.L.</td>
<td>B777-200ER, B777-200</td>
<td></td>
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<td>Flight attendant seat cushion foams replacement</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/authorised noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List, Testor document No. TAS-STC-B777PA-01-2014, Revision 1, dated 22 September 2014. - Master Drawing List, Testor document No. TAS-STC-B777PA-MWL-01 Revision N/A, dated 31 July 2014.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/10/2014</td>
<td>Active</td>
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<tr>
<td>10050757</td>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>REV. 1</td>
<td>TCCA STC S410-07, REV. 3</td>
<td>DHC-8-301/311-314/315, DHC-8-301/200, DHC-8-302/303-106</td>
<td>Installation of a Universal Avionics UNS-1 Lw Flight Management System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>See EASA Approved Annex</td>
<td>Antenna structural provisions provided under the latest applicable version of EASA STC MOT.21.NEU.1 are required. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/11/2014</td>
<td>Active</td>
<td>10050757</td>
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<tr>
<td>STC Holder</td>
<td>STC Holder</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>DART AEROSPACE Ltd.</td>
<td>TCCA SH10-48 ISSUE 4</td>
<td>R 44, R 44 8</td>
<td>CONFIGURATION A: Quick release mounting provision; Installation as defined in Aero Design Ltd DLCL065-1, Revision 1, dated 2 June 2014; ortate TCCA approved revision CONFIGURATION B: External Cargo Basket: Installation as defined in AeroDesign DLCL065-2, Revision 1, dated 2 June 2014, or later TCCA app...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>CONFIGURATION Aeronautical Design Instructions for Continued Airworthiness ICA 906.91, Revision 2, dated 27 May 2014, CONFIGURATION BII Aero Design Flight Manual Supplement FMS908.95, Revision 1, dated 27 May 2014</td>
<td>Prior to installation of this design change it must be determined that no adverse interaction between this design change and any other previously installed design change and/or repair will introduce road noise effect upon the airworthiness of the product.</td>
<td>10/11/2021</td>
<td>Active</td>
<td>10050756</td>
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<td>AIRGLAS, Inc.</td>
<td>FAA SR20431AK</td>
<td>R 44, R 44 8</td>
<td>Installation of the LAM-R44 Portable Device Mount</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Rotorscraft Flight Manual Supplement A/L-USOE-R44-FM, Rev. Orig., dated August 20, 2014, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/54/CF (subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU USA Biz...</td>
<td>Prior to installation of this design change it must be determined that no adverse interaction between this design change and any other previously installed design change and/or repair will introduce road noise effect upon the airworthiness of the product.</td>
<td>09/10/2014</td>
<td>Active</td>
<td>10050764</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>737-300, 737-500</td>
<td>TCAS II version 7.1 upgradeRev1: update of documentation listfle: update of documentation list</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Supplemental Type Definition Document STD304-14/19M rev 01/30/2016/30/2016, later revisions of the above listed documents approved by EASA.</td>
<td>- Pilots are authorized to deviate from their current ATC clearance to the extent necessary to comply with a TCAS II resolution advisory. When TCAS II voice message “CLEAR OF CONFLICT” is announced, the pilot must promptly return to the previous ATC clearance. - Maneuvers must not be based solely on info...</td>
<td>24/07/2015</td>
<td>Active</td>
<td>10050767</td>
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<tr>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>REV. 2</td>
<td>DHC-8-314, DHC-8-315</td>
<td>Installation of a Universal Avionics UNS-1Ew Flight Management System Revision 2: this revision was issued based on TCCA STC SA11-57, Issue 4dem adds an additional configuration to allow use of the MLS channel off the Avionics. Configurations 1 117366-402, 403, 402, 423, 424 and corresponding AFMS 117366-...</td>
<td>The Type and OS Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:Revation(s): F-103, Use of &quot;MLS” instead of &quot;LPV&quot;, issue 04The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>This change is defined with Universal Avionics UNS-1Ew FMS Document 117366-400, Issue 10, dated 10/09/2014, Universal Avionics UNS-1Ew FMS Document 117366-402, Rev. 6</td>
<td>This approval does not cover the structural provisions of the antenna installation. EASA STC 10059623 or another EASA approved antenna installation shall be installed prior to or in conjunction with the installation. Prior to installation of this change/repair it must be determined that no adverse interaction...</td>
<td>30/06/2022</td>
<td>Active</td>
<td>10050772</td>
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<tr>
<td>TASS-EU Ltd.</td>
<td>CAA UK RESPONSIBILITIY</td>
<td>A330-200</td>
<td>Passenger Seats - Refurbishment and Installation of IFE provisions</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:CS 25 Amdt 14: Various</td>
<td>TASS-EU LTD Master Document List No. 132192 Issue Original dated 03/12/2014, Revision 01/01/2014</td>
<td>Restricted to MSN 463 and 504</td>
<td>10/11/2014</td>
<td>Active</td>
<td>10050773</td>
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<td>GULFSTREAM AEROSPACE Ltd.</td>
<td>CAA UK RESPONSIBILITIY</td>
<td>GV-SP</td>
<td>Mid Cabin Pocket Door Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:Revation(s): CRI D-01 Installation of Interior Doors The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged...</td>
<td>Gulfstream AFS No. GA46, 14-1-AFS Revision - or later revisions of the above listed documents approved by EASA/Gulfstream Design Change Record No. GA46, 14-1-2-RCD, Revision New Dated 09 October 2014</td>
<td>EASA STC No. 10015929 is a pre-requisite for this STC</td>
<td>10/10/2014</td>
<td>Active</td>
<td>10050777</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A330-343/1-340-033</td>
<td>FANS A+ Over the Iridium Network.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document (ML): S21.MDL-1143 Issue 4. Minor Design Change S21.3d-71-1188</td>
<td>1. Airworthiness approval of this design change does not constitute operational or certification approval for FANS operation. The FANS system must be existing and must be approved. 2. Prior to the introduction of this design change it must be proven that the aircraft is provisioned with an ICS-220A dual.</td>
<td>22/05/2018</td>
<td>Active</td>
<td>10050778</td>
<td></td>
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<tr>
<td>PMV ENGINEERING</td>
<td></td>
<td>B737-600/-700/-800/-900</td>
<td>Mark II Aircraft Communications Addressing and Reporting System (ACARS) Installation</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Applicable paragraph(s) at a later amendment/CS-25 at amendment 10. The requirements for environmental protection and the associated noise and emissions levels are unaltered and remain applicable to this certificate/approval.</td>
<td>- Certification Investigation Report AVI-0008-CDR-R05 dated 10/10/2014 - Aircraft Flight Manual Supplement AVI-0008-AMRS-R02 dated 17/01/2011 - Instructions for Continued Airworthiness AVI-0008-ICA-R02 dated 02/2012 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>10/10/2014</td>
<td>Active</td>
<td>10050783</td>
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<td>B.A.S., INC.</td>
<td>REV. 1</td>
<td>FAA STC SA0185SE SEE EASA-APPROVED MODEL LIST</td>
<td>Installation of BAS, Inc. Shoulder Harness in accordance with the appropriate BAS Report/Drawings as listed on EASA Approved Model List (AML), or later approved revisions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with BAS Inc reports and drawings as defined in the attached AML list or, later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>13/10/2014</td>
<td>Active</td>
<td>10050784</td>
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<tr>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>REV. 1</td>
<td>B777-300ER</td>
<td>Cabin reconfiguration for LATAM in Brazil</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, EAD Aerospace document No. 01423-EAS-A-AML, Revision B(e) dated 23 October 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>05/11/2014</td>
<td>Active</td>
<td>10050787</td>
<td></td>
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<tr>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td></td>
<td>FAA STC ST0280CH 767-300</td>
<td>Installation for a Rockwell Collins SATCOM 2100 System - Configuration 1: For aircraft without existing Boeing SATCOM provisions - Configuration 2: For aircraft with existing Boeing SATCOM provisions with partial Boeing SATCOM provisions (HF/IFAT panel installed in P/Panel) - Configuration 3: For aircraft...</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements CS-25 amndt 6: 25.29(h)(2) -&gt; 25.30(h)(2) -&gt; 25.30(h)(1) -&gt; 25.30(h)(2) -&gt; 25.30(h)(3) -&gt; 25.30(h)(4) -&gt; 25.601(h) -&gt; 25.601(h)(9) -&gt; 25.611(h) -&gt; 25.613(h)...</td>
<td>Master Data List ECS-207397 rev For later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions thereof/decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement, and with EASA ED Decision 2004/04/CF (or subsequent revisions thereof/decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement.</td>
<td>- Limitations as per instructions for continued airworthiness ECS-207406 - FAA STC ST0280CH to be installed prior to or in conjunction with the changes of this certificate. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
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<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
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<td>FAA STC ST0288CH 767-300</td>
<td>Installation of an Intermediate Gain Antenna (IGA) subsystem - configuration 1: For aircraft without existing SATCOM antennaprovisions from Boeing - configuration 2: For aircraft with existing SATCOM antenna provisions from Boeing - configuration 3: Installation of IGA provisions for aircraft without...</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements CS-25 amndt. 6: 25.29(h)(2) -&gt; 25.30(h)(2) -&gt; 25.30(h)(3) -&gt; 25.30(h)(4) -&gt; 25.601(h) -&gt; 25.601(h)(9) -&gt; 25.611(h) -&gt; 25.613(h)...</td>
<td>Master Data List ECS-207305 rev For later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions thereof/decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement, and with EASA ED Decision 2004/04/CF (or subsequent revisions thereof/decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement.</td>
<td>As per instructions for continued airworthiness ECS-207306 Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of...</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>CAAA</td>
<td>1</td>
<td>RESPONSIBIL ITY</td>
<td>A330-200</td>
<td>IFE INSTALLATION STCInstallation of a new IFE system on a cabin configuration for Turkish Airlines Rev.1. Transfer of STC ownership from JAMCO AER DESIGN &amp; ENGINEERING ITCST TWENTY ONE LIMITED</td>
<td>The Certification Basis for the original product as amendability the following additional or alternative airworthiness requirements: Referred to section 7 in Certification Summary.</td>
<td>- MASTER DATA LIST: J0-40AC02-1, rev B-. ICA &amp; EWIS for Continued Airworthiness J0-40AC04-1, rev 1 Certification Summary Report JD-40AC10-1, rev A-. AFM Supplement: AFM-THY404-1 rev 2 or later revisions of the above listed documents approved by EASA.</td>
<td>This STC is concurrent and must to be embodied with STC 10050775Passenger seats refurbishment and Installation of IFE provisions) for aircraft installation. The following are required as pre-requisites for the installation of new STC EASASTC A01077 (Cabin modification)-Minor Change Modification...</td>
<td>24/05/2016</td>
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<td>DELTA ENGINEERING</td>
<td>FAA STC</td>
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<td>ST2093NY-O</td>
<td>A330-200-300</td>
<td>Installation of Panasonic Global Communication Suite</td>
<td>The Certification Basis for the original product as amendability the following additional or alternative airworthiness requirements: Refer to section 3 in Compliance Summary Report ref 6654-08051-2305 rev 0. The requirements for environmental protection and the associated certified noise and/or emissions levels of...</td>
<td>- MASTER DATA LIST: ref 0154-08031-2305, rev E-. ICA &amp; EWIS for Continued Airworthiness ref 0734-08026-2305(rev 0l, rev 0, rev 1, Compliance Summary Report Project results: ref 6654-08051-2305, rev B-. AFM Supplement: ref 0654-0807-2305, rev E-. FAA Letter &quot;EASA Approval of FAA Supplemental Ty...</td>
<td>1) Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce a random effect upon the airworthiness of the product. For The Airworthiness Limitations, refer...</td>
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<td>ISRAEL AEROSPACE</td>
<td>TCCA</td>
<td>1</td>
<td>6A276</td>
<td>737-900, 737-900ER 737-600, 737-700, 737-800 737-300, 737-400, 737-500</td>
<td>Rev 0: Towing airplane by towbarless tractor TaxiBot MB Rev 1: Inclusion of Boeing 737-700G Models</td>
<td>The Certification Basis for the original product as amendability the following additional or alternative airworthiness requirements: CRI A-01 Issue 4 dated 14.03.2016. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged...</td>
<td>AFMS: &quot;B737-300/400/500/600/700 700/800/900/900ER AIRPLANE FLIGHT MANUAL (AFM) SUPPLEMENT No. 1 - TAXIING USING TAXIBOT&quot;, Rev N., dated 05-02-16. (or subsequent revisions of the above listed documents approved by EASA: 737 Classic models (737-300/400/500) COMPLIANCE CHECK LIST (CCL). Ref. TXB0010/004. Rev F...</td>
<td>As per AFMS &quot;B737-300/400/500/600/700 900/900ER AIRPLANE FLIGHT MANUAL (AFM) SUPPLEMENT No 1 - TAXIING USING TAXIBOT&quot;, Rev. N. dated 05-02-16. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed...</td>
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<td>AEROTECH &amp; CONCEPT</td>
<td>MF900</td>
<td>1</td>
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<td>MF900</td>
<td>Self-polarizing Cabin Solar Protection System installation on MyaleFalcon 900 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AEROTEC Master Data List doc. N° AE13242AFMS Rev.0 dated 10.10.2014.Aircraft Flight Manual doc. N° AE13242AFMS issue 0 dated 18.06.2014Instructions for Continued Airworthiness doc. N° AE13242AFMS Issue 0 dated 20.02.2014 later revisions of the above listed documents approved by EASA.</td>
<td>Applicable to MHN 064 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a random effect upon the airworthiness of the product...</td>
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<td>AIRFILM CAMERA SYSTEMS</td>
<td>FAA STC</td>
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<td>SR02453A</td>
<td>BELL 429</td>
<td>Nose Utility Mount Light W3 AFM-MM025K-X-1. The light kit consists of two mounting brackets and two Telodyne LEDHalo bulbs landing light assemblies. This installation is eligible for Bell 429 helicopters with Airfilm Camera System STC 10039428 installed.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airworthiness Limitation Section of Instructions for Continued Airworthiness AFM-MM025K-008, Rev. INC., dated August 25, 2012. or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/19/CF (or subsequent revisions off this decision) and/or the Technical Im...</td>
<td>This installation is only eligible for Bell 429 robotic aircraft equipped with Airfilm Camera System EASA STC number 10039428. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair...</td>
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<td>10050842</td>
<td>FIELD AVIATION EAST, LTD.</td>
<td>REV. 2</td>
<td>TCGA O-LSA13-137/D</td>
<td>DHC-8-311</td>
<td>Universal UNS-1E(w) and UNS-1F(w) Flight Management System Installation. The installation options are based upon converting older single or dual-lugacy equipment with the latest technology. Revision 1 of this STC was issued to include LPV operation (SN 592) (CRIF-102, issue 2). Revision 2 of this STC was...</td>
<td>DHC-8-311 UNS-1E(w) Installation (SN 407, SN 493, SN 494, SN 532, SN 535, SN 585, UNS-1E(w) (SW, SCN 1000.x) Installation must be performed in accordance with FACI Master Drawings Data List designated D585/0000. Rev A Flight operations must be performed in accordance with FACI Flight manual Supplement designated D585/3000.</td>
<td>Limited to SN 238, SN 238, SN 407, SN 493, SN 494, SN 532, SN 535, SN 585, UNS-1E(w) Flight Management Systems approved for LPV. Route, Terminal and Non-Precision approach operations only. The installation of Honeywell Sympo...</td>
<td>09/10/2019</td>
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<td>10050847</td>
<td>SWS CERTIFICATION SERVICES LTD.</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>UCS3 Berth Activation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate to the certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>ICA, Maintenance Manual Supplement 3132-MMS-01 Issue 2 (Design Report 3132-DR-01 Issue 1) Master Configuration Scheme ref 3132-MCS-01 Issue 1 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
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<td>10050848</td>
<td>FLYBE LIMITED</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>DHC-8-402, DHC-8-401 Installation of Tripospheric Airborne Meteorological Data Reporting (TAMAR) System, DHC-8-400</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Installation Must be in accordance with Flybe Engineering Order C2132-002. Revision A, Operation must be in accordance with Flybe AFM Supplement RE2112-18, Issue 1. Continued Airworthiness shall be in accordance with the FAA or later revisions of the above listed documents approved.</td>
<td>None</td>
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<td>SWS CERTIFICATION SERVICES LTD.</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>STG Aerospace Saftgts Photoluminescent Floor Proximity Emergency EscapePath Marking System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25 Amendment 14 varies the requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and...</td>
<td>SW5 Certification Services Ltd Airline Flight Manual Supplement No. 3133-1 FMS-01 Issue 1 dated 29 September 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
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<td>10050851</td>
<td>IAC - INDUSTRIAS AERONAUTICAS DECOINBRA, Ltda.</td>
<td></td>
<td>CESSNA 206 SERIES</td>
<td>Modification for skydiving operations includes a sliding door, the removal of the co-pilot control wheel and the introduction of an instrument panel protection, a static line parachutes support, and a parachute’s platform fixed to the wheel nutand in the foot step of the right landing gear leg.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25 Amendment 14 Parachute Activity The requirement for environmental and the associated certified noise and/or emissions levels of the original product are unchanged...</td>
<td>APM Supplement For Parachuting Modification VAC-202-APM-ED for later revisions of the above listed documents approved by EASA. IAC-C206 MLD Master Document List, Issue 1 11-02-2014 FORM IAC21.001 Project Dossier Issue 1 18-01-2014AIC- C206 Annex I Manufacturing Issue 1 18-01-2014AIC-C206 Annex II Op...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/10/2014</td>
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## List of EASA Supplemental Type Certificates

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<td>10050855</td>
<td>NORTWEST AEROSPACE TECHNOLOGIES NC</td>
<td>A321-231</td>
<td>REV. 2</td>
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<td>Cabin Interior Reconfiguration of British Airways aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10050856</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
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<td>Fuselage cutouts in Cessna 172</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition 3 (a) and 3 (b) of the Certification Basis for the original product, Fuselage cutouts in Cessna 172.</td>
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<td>10050861</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>B300C</td>
<td>REV. 5</td>
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<td>Installation of MTV-27-1-( )/C-F-R/P/PR260-65( ) Propeller on Beechcraft King Air B300C...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition 3 (a) and 3 (b) of the Certification Basis for the original product, Fuselage cutouts in Cessna 172.</td>
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<td>H4 AEROSPACE (UK) Ltd</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>747-8F</td>
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<td>Honeywell TCA9 7.1 System Upgrade on Boeing 747 Aircraft for Honeywell Systems</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>AIM AVIATION (HENDHALLS) LTD</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>757-300</td>
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<td>Galley 4A in B757-300 for Condor</td>
<td>The Certification Basis (CB) for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition 3 (a) and 3 (b) of the Certification Basis for the original product, Fuselage cutouts in Cessna 172.</td>
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<td>Installation, Indium Antenna (Structural Provision)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>HAECK CABIN SOLUTIONS, LLC</td>
<td>ST04168AT</td>
<td>FAA STC NO. ST04168AT</td>
<td>767-300</td>
<td>Interior Reconfiguration of a Boeing 767-300 Aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC NO. ST04168AT dated 17 October 2014. TAMCO Aeronautics Master Data List 13T448L001, Revision A, dated October 17, 2014.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>GE AVIATION SYSTEMS L.L.C.</td>
<td>REV. 1</td>
<td>AS 350 BB, AS 350 B2</td>
<td>727-200, 727-200C 727-100, 727-100C</td>
<td>Installation of a Smiths Industries Combined Voice and Flight Data/Record</td>
<td>The Certification Basis for the original product as amended by the following additional requirements:</td>
<td>AFM Supplement, Smiths Industries Doc 1Y183 Rev X1, FAA approved 29/08/2002. Smiths Industries Drawing No. 176879, Revision B, dated July 25, 2002</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>AS 350 BB, AS 350 B2</td>
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<td>Window and/or Ejection Door (ESD) Protection System installed upon the airworthiness of the product.</td>
<td>The Certification Basis for the original product as amended by the following additional requirements:</td>
<td>Flight Manual Supplement for Wire Strike Protection System in AS350BB/doc. no. 35032-500206 Ed. 03 dated 24/09/2015, or later revisions of the above listed documents approved by EASA.</td>
<td>For rotorcraft model AS 350 B2 this STC is only applicable if Eurocopter SR 53.00.39 is installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
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<td>A109E, A109S</td>
<td>Installation of Dual Garmin GTN 650</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Approved Flight Manual Supplement for COMNAV/GS/GPS Garmin/GTN650, ASR-2014-004-GPSGS-050-00-00 dated 16.10.2014, A109E/11102 - 11077 with Emergency Builder kit; S/N 11078 and above, A109S all S/N Approved Flight Manual Supplement for COMNAV/GS/GPS Garmin/GTN650, ASR-2014-004-FMS-050-00-00 dated 16.10.2014.</td>
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<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Approved Flight Manual Supplement for COMNAV/GS/GPS Garmin/GTN650, ASR-2014-004-GPSGS-050-00-00 dated 16.10.2014. Approved Flight Manual Supplement for COMNAV/GS/GPS Garmin/GTN650, ASR-2014-004-FMS-050-00-00 dated 16.10.2014.</td>
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<td>27/10/2014</td>
<td>Active</td>
<td>10050930</td>
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<td>EASA STC no</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Certification basis for original product</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10050942</td>
<td>PMV ENGINEERING</td>
<td>REV. 2</td>
<td></td>
<td>A321-211/-212/-213/-231-232, A321-111/-112/-131, A320-231, A321-211/-212/-233, A320-211/-212/-213/-215/-216, A319-111/-112/-133, A319-111/-112/-114/-115, A318-111/-112/-121/-122</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Document List AVI-00669-MLR-R2, dated 05 Oct. 2017</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/10/2017</td>
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<tr>
<td>10050952</td>
<td>SAFRAN ELECTRONICS &amp; DEFENSEBUSINE SE UNT. HELICOPTERS DATAMANAGEMENT NT SOLUTIONS &amp; SERVICES</td>
<td>REV. 1</td>
<td></td>
<td>BO105 LS A-1, BO105 LS A-3, BO105 A, BO105 C, BO105 D</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s): CNF F-61 Lithium Battery Installation Date: October 2nd, 2014, The requirements for the installation of the HELICOM V2 ANA Flight Data Monitoring and Transmitting system. The proposed change described in ISEI document N3-BO105-HV2-ANA and consists of two possible configurations. The first configuration comprises PN4720F and PN4730F, the second configuration comprises PN4720P and PN4740P.</td>
<td>Rotorcraft Flight Manual Supplement FMN 011-BO105-HV2 ANA-E-B-02, Rev. 02, dated October 27th, 2014, Installation Manual NT-BO105-HV2 ANA-E-B-02, Rev. 02, dated October 27th, 2014 Instruction and Compliance Report TEN 714/47 Issue 2.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/02/2020</td>
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<td>10050956</td>
<td>ARCHEION HOLDINGS, L.L.C.</td>
<td></td>
<td>FAA STC ST089B1AT</td>
<td>B727-200/-200F</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s): (H=01) ICA on EWS, the following paragraph(s) at a later amendment: CS25 and 4: 25.29, 25.301, 25.303, 25.305, 25.307, 25.309, 25.311, 25.313, 25.315, 25.317, 25.319, 25.401,...</td>
<td>Master Data List 080920-001-MLD Rev 0, Elecrane Flight Manual Supplement 080920-503 Rev 0, and Customer Flight Manual Supplement 080920-504 Rev 0, Weight and Balance Change Report 080920-910 Rev A or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/...</td>
<td>As per Instructions for Continued Airworthiness 080920-504 last version.</td>
<td>28/10/2014</td>
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<td>10050956</td>
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<td>10050977</td>
<td>CESSNA DUESSELDORF CITATIONBERVIC E CENTER GmbH</td>
<td></td>
<td>CESSNA 680 - CITATION SOVEREIGN</td>
<td>Installation of a Single UNS-1Lw Universal Avionics Flight Management System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Declaration of Compliance, Document of Cessna Düsseldorf Citation Rev. 2 dated 24.10.2014 Master Document List Ref. 2014-6-16 dated 16.10.2014 Engineering Order Ref. 2014-6 dated 16.10.2014 later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>28/10/2014</td>
<td>Active</td>
<td>10050977</td>
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<tr>
<td>10050967</td>
<td>TENENCIA Ltd.</td>
<td></td>
<td>CAA UK RESPONSIBILITY MARK 050 AND 0502</td>
<td>Installation of a Single UNS-1Lw Universal Avionics Flight Management System with LPV</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Modification Report Tenencia report TEN 714 Issue 4, Certification Compliance Report Tenencia report TEN 714/1 Issue 2, Flight Manual Supplement Tenencia report TEN 714/4 Issue 4, Instructions for Continued Airworthiness Tenencia report TEN 714/4 Issue 3, later revisions of the above listed documents approved by EASA.</td>
<td>As defined in Flight Manual Supplement Tenencia report TEN 714/4 Issue 3, prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/10/2014</td>
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<td>10050967</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AIRCAD</td>
<td>A321-213</td>
<td>Installation of two lower L-band ATG (P/N 2439-89-03) antennas and ansopix GPS (P/N C17065AA) antenna</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Compliance Check List CCL-13010 Rev.1, dated 01/10/2014; Master Data List MDL-13010 Rev.1, dated 14/10/2014; Instructions for Continued Airworthiness ICA-13010 Rev.1, dated 14/10/2014; or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>29/10/2014</td>
<td>Active</td>
<td>10050970</td>
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<td>ASI INNOVATION</td>
<td>ERJ 170-200 LR</td>
<td>EMBRAER 175 CABIN MODIFICATION</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>STC Data List ref. ASI-S6D-042-02, dated 22/10/2014 or later revisions of the above listed documents approved by EASA.</td>
<td></td>
<td>29/10/2014</td>
<td>Active</td>
<td>10050975</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>B757-200</td>
<td>B757 WiFi activation for Vision Systems (FE suite). This modification consists in the activation of a WiFi System on a/b757-200. The system provides airborne network server access to the aircraft's cabin passengers via Portable Electronic Devices (PED) Such devices could include laptop computers and WiF...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Master Data List AEROTEC n° AE14100MDL rev.1 dated 28/10/2014 or later revisions of the above listed documents approved by EASA.</td>
<td>1. To install this modification, the aircraft must have the following minor modification already embodied (prerequisite configuration): AEROTEC AE14098.2.</td>
<td>29/10/2014</td>
<td>Active</td>
<td>10050984</td>
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<td>BOMBARDIER INC.</td>
<td>BD-700-1A11</td>
<td>Installation - Upgrade from Swift64 to SwiftW-band (EAS/Avaluation of Transport Canada's Supplemental Type Certificate/SA14-52 at issue 2</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Modification to be completed in accordance with Transport Canada approved EMTEC Canada Ltd. Modification Data Summary Number J686000Rev N/C. For instructions for continued airworthiness, refer to J686070-ICARerevision N/C or later revisions of the above listed documents approved by EASA inaccordance.</td>
<td>None.</td>
<td>29/10/2014</td>
<td>Active</td>
<td>10050986</td>
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<td>DELTA G DESIGN Inc.</td>
<td>TCCA STC SA14-52 ISSUE 2</td>
<td>737-700/700-800/900/-900ER 737-300/400-500/-600 Structural modification necessary for the installation of actual IntrumAntenna.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CS22-Andtl. 0: 25.571(a)(b), 25.613, 25.149(a) CS25Andtl. 9: 25.28(3)-25.305(a)(b), 25.30(7) 25.28(1)(c)(d), 25.30(1)(c), 25.30(3), 25.30(4a), 25.34(1a), 25.35(1), 25.35(1) or latest</td>
<td>Master Data List DGD-RU002-MD-01 rev C Instructions for Continued Airworthiness DGD-RU002-MD-01 rev C or later revisions of the above listed documents approved by EASA inaccordance.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other concurrently installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/10/2014</td>
<td>Active</td>
<td>10050987</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST09913NY</td>
<td>A319, A320, A321 Structural modification necessary for installation of a Panasonic Avionics Corporation In-Flight Entertainment and/or Global Communications Suite (GCS) System/Export Darnot option) The Revision 1 extends the STC applicability for Aircraft A321, and provide the eXphone functionality. The Revision 2 updates the Master Data List 0164-07820-2301 for...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CS22-Andtl. 0: 25.571(a)(b), 25.613, 25.149(a) CS25Andtl. 9: 25.28(3)-25.305(a)(b), 25.30(7) 25.28(1)(c)(d), 25.30(1)(c), 25.30(3), 25.30(4a), 25.34(1a), 25.35(1), 25.35(1) or latest</td>
<td>Delta Engineering Master Data List 0164-07820-2301, Revision BB, dated 25/04/2019, or later FAA approved revisions. The Instructions for Continued Airworthiness and Airplane Flight Manual Supplement, as issued by the Master Data List, are required with this installation. 1. Operational approval to use T-PED's must be obtained separately from the STC. 2. FAA STC ST09809DY (Antenna) is required with this STC when installed</td>
<td>02/08/2019</td>
<td>Active</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Model(s)</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>SABENA TECHNICS BGC</td>
<td>Full Cabin Modification with Zodiac Inflight Innovations (ZII) Rave IFE Installation/Upgrade on A320-200 This STC installs the Rave IFE system, which mainly consists in the transformation through the current cabin arrangement of Zodiac STC, Seat power unit, Seat Display Unit, Crew Panel and Seat P...</td>
<td>A330-243</td>
<td>CS 25 Amendment 4, except for 25.562 and 25.765 for which the certification basis specified in EASA TCDS A.004 issue 36 applies</td>
<td>For detailed applicable requirements refer to the Certification Plan ref: 4633-02-A-2101-R02 section 3.4 The requirements for environmental protection and the associated tests...</td>
<td>Master Data List ref 4633-02-A-2103-R00, dated 27 October 2014</td>
<td>30/10/2014</td>
<td>Active</td>
<td>10051003</td>
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<td>ECMS AVATION SYSTEMS GMBH</td>
<td>External Load System on EC155 Load systems, installed on the floor-down support threaded in the helicopter frame, used as hard points.</td>
<td>EC 155 B1, EC 155 B</td>
<td>CS 29-CS 29.25, CS 29.27, CS 29.45, CS 29.65, CS 29.141, CS29.341, CS 29.251, CS 29.301, CS 29.293, CS 29.309, CS 29.307, CS 29.309, CS 29.321, CS29.337, CS 29.341, CS 29.571, CS 29.603, CS 29.605, CS 29.609, CS29.616, CS 29.611, CS 29.613, CS 29.625, CS 29.865, CS 29.1301, CS29.1309, CS 29.1323, CS...</td>
<td>Flight Manual Supplement &quot;External Load System EDX07 on EC155&quot;, Original Issue dated 1st October 2014 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the integration between these systems, installed on the tie-down support threads of the helicopter frame, used as hard points.</td>
<td>03/11/2014</td>
<td>Active</td>
<td>10051007</td>
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<td>FOKKER SERVICES B.V.</td>
<td>TCAS Software Upgrade to TCAS V7.1 New Design Change to STC 10051008 with the ACS5 TCAS II equipped aircraft the eligibility of the STC</td>
<td>A321-231, A321-232, A321-212, A321-213, A321-112, A321-131, A321-211, A320-232, A320-233, A321-111, A320-215, A320-216, A320-221, A320-211, A320-212, A320-214, A319-131, A319-133, A319-132, A319-134, A319-111, A319-112, A318-112, A318-121</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval</td>
<td>Certification Plan CP-0266 Issue 1</td>
<td>Compliance Report CRR-0266 Issue 1 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the integration between this product and any other previously installed design change and/or repair will introduce noise and/or vibration effect upon the airworthiness of the product.</td>
<td>22/04/2015</td>
<td>Active</td>
<td>10051008</td>
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<td>PW Inc/D&amp;B/A PW6</td>
<td>Installation of Cabin LED Downwash Lighting.</td>
<td>FAA STC SA0174WI</td>
<td>CS 25 Amendment 4, except for 25.562 and 25.765 for which the certification basis specified in EASA TCDS A.004 issue 36 applies</td>
<td>Installation of Cabin LED Downwash Lighting in accordance with PWIMaster Document List, No. MOL910080 Rev. IR dated 13 April 2015 (v.+.) &gt; if later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/ACF (or subsequent revisions of this document) and/or...</td>
<td>Prior to installation of this design change it must be determined that the integration between this product and any other previously installed design change and/or repair will introduce noise and/or vibration effect upon the airworthiness of the product.</td>
<td>31/10/2014</td>
<td>Active</td>
<td>10051015</td>
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<td>NORTHERN AVIONICS S.R.L.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements EASA CS-ACS (Subpart B) The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and rem...</td>
<td>A108A</td>
<td>NAV-490 &quot;Installation of Garmin GTN 650 COM/NAVAIP/OP/SAS Navigation System&quot;</td>
<td>Rotorcraft Flight Manual Supplement RFMS NAV-490, Rev. 0, dated 22.10.2014 for later revisions of the above listed documents approved by EASA Certification Program (includes Certification Basis), NAV-490, Issue C, dated 30.07.2014 Modification Bulletin NAV-490, Rev. 0, dated 29.09.2014 (includes Drawing L...</td>
<td>Prior to installation of this design change it must be determined that the integration between this design change and any other previously installed design change and/or repair will introduce noise and/or vibration effect upon the airworthiness of the product.</td>
<td>31/10/2014</td>
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<td>10051016</td>
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<td>EASA STC no.</td>
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<td>Rev. level/STC ref.</td>
<td>Foreign/NAA STC ref.</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10051018</td>
<td>MECAER AVIATION GROUP S.p.A.</td>
<td>REV. 4</td>
<td>6E4M0000001-001</td>
<td>MBB-BK117 D-3 MBB-BK117 D-2</td>
<td>Installation of EMS interiors P/N 6E4M0000001-001 Revision 1 is to add the installation of a side facing seat bench on the right-hand side of the cabin. Revision 2 is to extend the applicability to MBB-BK117 D-3. Revision 3 is to close the post-approval action item related to the installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Declarative of Compliance document number NDC 6E4M0000001-001 Revision 1, dated 31.08.2023.</td>
<td>Certification basis, Declaration of Compliance document number NDC 6E4M0000001-001.</td>
<td>20/02/2023</td>
<td>Active</td>
<td>10051018</td>
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<td>10051023</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC Q-LSAO2-071/D ISSUE 1</td>
<td>6E4M0000001-001</td>
<td>BD-700-1A10</td>
<td>Miscellaneous Cabin Modifications, EASA validation of Transport Canada's Limited Supplemental Type Certificate Q-LSAO2-071/D at issue 1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Declarative of Compliance document number NDC 6E4M6000001-001.</td>
<td>Certification basis, Declaration of Compliance document number NDC 6E4M6000001-001.</td>
<td>31/10/2014</td>
<td>Active</td>
<td>10051023</td>
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<td>10051024</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC Q-LSAO5-053/D ISSUE 1</td>
<td>6E4M0000001-001</td>
<td>BD-700-1A10</td>
<td>Miscellaneous modifications of the cabin electrical system. EASA validation of Transport Canada's Limited Supplemental Type Certificate Q-LSAO5-053/D at issue 1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Declarative of Compliance document number NDC 6E4M4000001-001.</td>
<td>Certification basis, Declaration of Compliance document number NDC 6E4M4000001-001.</td>
<td>31/10/2014</td>
<td>Active</td>
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<td>10051026</td>
<td>ASI INNOVATION</td>
<td>REV. 1</td>
<td>6E4M0000001-001</td>
<td>ERU 170-200 LR</td>
<td>SATCOM System Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Declarative of Compliance document number NDC 6E4M2000001-001.</td>
<td>Certification basis, Declaration of Compliance document number NDC 6E4M2000001-001.</td>
<td>05/11/2014</td>
<td>Active</td>
<td>10051026</td>
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<td>10051041</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC ST04125SE</td>
<td>777-200</td>
<td>Installation of Interior Reconfiguration for KLM</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25.6313 Amdt 1CS 25.853(c)(3)(i)(b) Amdt 36E4M0000001-001 Revision 1</td>
<td>The requirements for environmental protection and the associated certification are unchanged and remain applicable to this certificate/ approval.</td>
<td>Declarative of Compliance document number NDC 6E4M2000001-001.</td>
<td>Certification basis, Declaration of Compliance document number NDC 6E4M2000001-001.</td>
<td>04/11/2014</td>
<td>Active</td>
<td>10051041</td>
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### List of EASA Supplemental Type Certificates

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<th>STC No.</th>
<th>STC Holder</th>
<th>Rev. Level</th>
<th>Foreign/NAAR STC Ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification Basis</th>
<th>Associated Technical Documentation</th>
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<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>10051047</td>
<td>MID-CANADA MOD CENTER</td>
<td></td>
<td>TCCA STC O-LSA07-123/D ISSUE 4</td>
<td>F50EX</td>
<td>Dual Universal UNS-1 F FMS, Honeywell MK V Enhanced EGPWS and AirCell Axxess Satphone Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Avionics Design Services Ltd. Master Drawing List MDL07041 REV. F dated 17.10.2014 Mid Canada Mod Center Airplane Flight Manual Supplement N° FMS07041-1Rev. B dated 10.10.2014 (DualUNS-IF FMS)Mid Canada Mod Center Airplane Flight Manual Supplement N° FMS07041-2Rev. B dated 10.10.2014 (Honeywell MK-V)</td>
<td>MSN 278</td>
<td>04/11/2014</td>
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<tr>
<td>10051055</td>
<td>H4 AEROSPACE (UK) Ltd</td>
<td></td>
<td>CAI UK RESPONSIBIL ITY</td>
<td>717-200</td>
<td>TCAS System Upgrade to Change 7.1 on Boeing MD 90-30/B717-200 series aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>H4 Aerospace Document No H4AFMS0542 Rev IR - Airplane Flight Manual Supplement or later revisions of the above lists of documents approved by EASA. H4 Aerospace Document No H4AFMS0749 Issue 3 - Modification Summary SHEET and Declaration of Compliance Change Approval Sheet (including Master Data List), Air France reference CAS- B777-14-006 revision 00 or later revisions of the above lists of documents approved by EASA.</td>
<td>This modification is limited to the Boeing 717-200 series aircraft/based in H4 Aerospace Modification Summary H4A1749Issue 3. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/11/2014</td>
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<tr>
<td>10051063</td>
<td>SOCIETE AIR FRANCE S.A</td>
<td></td>
<td>777-300, 777-300ER, 777-200, 777-200LR</td>
<td>G700 Electronic Flight Bag activation on B777</td>
<td>This modification consists in activating, on Boeing 777 aircraft, an Electronic Flight Bag system based on two 10-inch displays from UTAS: the G700 SmartDisplay Units (SDU). The G700 EFB is activated as a &quot;portable&quot; device as per the definition in EASA AMC 20...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Change Approval Sheet (including Master Data List), Air France reference CAS- B777-14-006 revision 00 or later revisions of the above lists of documents approved by EASA.</td>
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<td>10051064</td>
<td>INNOTECH AVIATIONNA DIVISION OF IMP GROUP LTD</td>
<td></td>
<td>TCCA O-LSA10-019/D ISSUE 1</td>
<td>BD-700-1A10</td>
<td>Passenger Cabin Entertainment System Upgrade EASA validation of Transport Canada's Supplemental Type Certificate O-LSA10-019/D at issue 1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Modification must be performed in accordance with the following: 1. JMLD30107020 Master Document List - Passenger Cabin Entertainment System Upgrade, BD-700-1A10, SN 9036, revision N/C, dated 05th March 2010, and 2. TL30107-004, Time Limits and Maintenance Checks - Cabin Entertainment System Upgrade, A...</td>
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<td>10051065</td>
<td>Indium Phone System Installation EASA validation of Transport Canada’s Limited Supplemental Type Certificate G-LSA02-0670 at issue 3.</td>
<td>The Installation must be done in accordance with: 1. MSDL0166004 (v2) Aerocas Airworthiness Limitations Mandatory Maintenance Manual Supplement, 2. AC60-166009.</td>
<td>The installation must be done in accordance with: 1. MSDL0166004 (v2) Aerocas Airworthiness Limitations Mandatory Maintenance Manual Supplement, 2. AC60-166009.</td>
<td>See those of MMS02166-009, Aerocas Airworthiness Limitations Mandatory Maintenance Manual Supplement, AirCell Indium Antennal Installation, Aircraft Serial Number 9036, Revision(MC), dated 07/24/2014. This STC installation is limited to aircraft serial number 9036.</td>
<td>06/11/2014</td>
<td>Active</td>
<td>10051065</td>
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<td>10051077</td>
<td>Installation of Flight Data Monitoring and Transmitting System HELICOMV ANA or HELICOM 1201+ ANA Revision 1: STC holder name change from I.S.E.I International Services to SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT - HELICOPTERS DATAMANAGEMENT SOLUTIONS &amp; SERVICES.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v): Cert F-01 Issue 3 Lithium Battery Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v): Cert F-01 Issue 3 Lithium Battery Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>CERTIFICATION PROGRAM NB-8K117 HV-JRA, Edition B Rev. 2 Installation Manual N7-BK117-HV-JRA Edition B, Rev. 2.0 Additions for Continued Airworthiness NB-8K117-HV-JRA Edition B. Rev. 02.</td>
<td>None</td>
<td>27/02/2020</td>
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<td>10051078</td>
<td>Installation of Flight Data Monitoring and Transmitting System HELICOMV ANA or HELICOM 1201+ ANA Revision 1: STC holder name change from I.S.E.I International Services to SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT - HELICOPTERS DATAMANAGEMENT SOLUTIONS &amp; SERVICES.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v): Cert F-01 Issue 3 Lithium Battery Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v): Cert F-01 Issue 3 Lithium Battery Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement VHR Aerospace No. V407-0822-FMS Rev. 0. Objections for Continued Airworthiness VHR Aerospace No. V407-0822-ICARev.0 for later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement Mast..</td>
<td>None</td>
<td>07/11/2014</td>
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<td>10051081</td>
<td>Installation of Flight Data Monitoring and Transmitting System HELICOMV ANA or HELICOM 1201+ ANA Revision 1: STC holder name change from I.S.E.I International Services to SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT - HELICOPTERS DATAMANAGEMENT SOLUTIONS &amp; SERVICES.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v): Cert F-01 Issue 3 Lithium Battery Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v): Cert F-01 Issue 3 Lithium Battery Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Instructions for continued airworthiness: No. 3499-D15, dated 06/10/2014. Airplane Flight Manual Supplement; No. 3499-D11, dated 06/10/2014. for later revisions of the above listed documents approved by EASA Master Document List No. 3499-017, dated 06/10/2014.</td>
<td>None</td>
<td>07/11/2014</td>
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<td>10051096</td>
<td>Installation of Flight Data Monitoring and Transmitting System HELICOMV ANA or HELICOM 1201+ ANA Revision 1: STC holder name change from I.S.E.I International Services to SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT - HELICOPTERS DATAMANAGEMENT SOLUTIONS &amp; SERVICES.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v): Cert F-01 Issue 3 Lithium Battery Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v): Cert F-01 Issue 3 Lithium Battery Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The approval holder shall fulfill the obligations of Part 21 §21.118 Prior to Installation of this design change it must be determined that the change is compatible with the original product as amended and is compatible with any existing change to the original product that has been approved. The requirements for environmental protection are unchanged and remain applicable to the original product. The approval holder shall fulfill the obligations of Part 21 §21.118 Prior to Installation of this design change it must be determined that the change is compatible with the original product as amended and is compatible with any existing change to the original product that has been approved. The requirements for environmental protection are unchanged and remain applicable to the original product.</td>
<td>None</td>
<td>07/11/2014</td>
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<td>10051113</td>
<td>Installation of Flight Data Monitoring and Transmitting System HELICOMV ANA or HELICOM 1201+ ANA Revision 1: STC holder name change from I.S.E.I International Services to SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT - HELICOPTERS DATAMANAGEMENT SOLUTIONS &amp; SERVICES.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v): Cert F-01 Issue 3 Lithium Battery Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v): Cert F-01 Issue 3 Lithium Battery Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement VHR Aerospace No. V407-0822-FMS Rev. 0. Objections for Continued Airworthiness VHR Aerospace No. V407-0822-ICARev.0 for later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement Mast..</td>
<td>None</td>
<td>07/11/2014</td>
<td>Active</td>
<td>10051013</td>
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**Notes:**
- **EC 135 T3**
- **EC 135 T2**
- **EC 135 T2**
- **EC 135 T1**
- **EC 135 P2**
- **EC 135 P1**

**References:**
1. Airbus Service Bulletin MSN 1892 and MSN 1902
2. Aircraft serial numbers: MSN 1892, MSN 1902, MSN 1663, MSN 1672, MSN 1771, MSN 1804, MSN 1834, MSN 1892 and MSN 1902
3. Airbus Service Bulletin A320-25-1 B42 must be incorporated concurrently with this STC.
4. Prior to installation of this design change it must be determined that the change is compatible with the original product as amended and is compatible with any existing change to the original product that has been approved. The requirements for environmental protection are unchanged and remain applicable to the original product.
5. Prior to installation of this design change it must be determined that the change is compatible with the original product as amended and is compatible with any existing change to the original product that has been approved. The requirements for environmental protection are unchanged and remain applicable to the original product.
6. Prior to installation of this design change it must be determined that the change is compatible with the original product as amended and is compatible with any existing change to the original product that has been approved. The requirements for environmental protection are unchanged and remain applicable to the original product.
7. Prior to installation of this design change it must be determined that the change is compatible with the original product as amended and is compatible with any existing change to the original product that has been approved. The requirements for environmental protection are unchanged and remain applicable to the original product.
8. Prior to installation of this design change it must be determined that the change is compatible with the original product as amended and is compatible with any existing change to the original product that has been approved. The requirements for environmental protection are unchanged and remain applicable to the original product.
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<th>STC Holder</th>
<th>FAA STC Level</th>
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<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>DUNGAN AVIATION, INC.</td>
<td>FAA ST1648W-0</td>
<td>FALCON 2000, FALCON 2000EX</td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List ref. 110067103 revision 0, dated 12 February 2013. AFM Supplement ref. 110067108 revision A, dated 30 October 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/4/C (or subsequent revisions of either decision) and/or L...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/11/2014</td>
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<td>ICELANDAIR EHF</td>
<td>REV. 1</td>
<td>757-200-300 767-300</td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) in a later amendment: -Type Certification Basis for parts not changed or not affected by this change.</td>
<td>- Icelandair Master Drawing Lists ref. DWG-767-2320-19 Initial Release and/or AMF-767-2320-60 Revision A - Icelandair Instructions for Continuing Airworthiness ref. CAI-767-2320-19 L.R. and CAI-757-2320-60 Rev...</td>
<td>Aeroplanes modified with STC must be - Operated in accordance with Icelandair AMF supplements ref AMF-767-2320-19 Initial Release or AMF-757-2220-60 Revision A respectively. - Maintained in accordance with Icelandair Instructions for Continuing Airworthiness ref. CAI-767-2320-19 L.R. or CAI-757-2320-60 Rev...</td>
<td>28/11/2016</td>
<td>Active</td>
<td>10051148</td>
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<tr>
<td>KLM ENGINEERING &amp; MAINTENANCE</td>
<td>BOEING 777-200</td>
<td></td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI-D-245 (Post TPC special condition - a/Overhead Crew Rest Compartment(s)) closed on 16 April 2004, which have adopted special condition FAA 25-230 SC Th...</td>
<td>Master Data List, KLM document No. MDL-612995, Revision 8, dated 12 November 2014 applicable revisions for EASA or later revisions of the above listed documents approved by EASA.</td>
<td>Other On 777-200 MSN 33712 with seat P/N 37104740-40-031, S/N 601598 and seat P/N 76710470-40-051, S/N 601600, the use of the Overhead Flight Attendant Rest (OFAR) is prohibited unless Recaro Service Bulletin 710.25-02 has been accomplished or armrest P/N 470-00-30-01, P/N 470-00-30-02, P/N 470-00-3...</td>
<td>12/11/2014</td>
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<td>10051152</td>
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<td>AVIATION PARTNERS, INC.</td>
<td>FAA STC ST02241SE</td>
<td>Installation of Aviation Partners Inc. Bended Winglets in the Dassault Aviation Mystere 50.</td>
<td></td>
<td>See Appendix This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>Aviation Partners, Inc. (API) Master Drawing List ref. APFS-0601, rev.A, dated 30 July 2009, AVIATION Partners, Inc. (API) AFM Supplement ref. APFS-0601 Cota 001 rev. 1, dated 14 March 2013. Aviation Partners, Inc. (API) Airworthiness Limitations as recorded in document ref. APFS-0601, rev. B, dated...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/11/2014</td>
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<td>SOCIETE D'ETOILE ET DE GESTION AERONAUTIQUE AUTIQUE (SÉGA)</td>
<td>AS355 E, F, F1, F2, N</td>
<td>Installation of ELT KANNAD 406 AF-Compact.</td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement H-13-01. KANNAD ELTAS355 Ed. 1 Nov 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/11/2014</td>
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## List of EASA Supplemental Type Certificates

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<th>Status</th>
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<td>10051169</td>
<td>NSE</td>
<td>REV 1</td>
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<td>EC135 T1, EC135 T2, EC135 T2+</td>
<td>Data Transmission System DTS KIT EC135 T1, EC135 T2, EC135 T2+ to EC135 T1, P1, EC135 P2, EC135 P2+ System is a Gecelization system to transmit aircraft's position, speed, altitude, and bearing</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Rutoncraft Flight Manual Supplement RFMS-DTSV2-EC135 Rev. 00 or later revisions of the above listed documents approved by EASA. A340-600, A340-600, A340-600</td>
<td>20/02/2015</td>
<td>Active</td>
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<tr>
<td>10051171</td>
<td>LUFTHANSA</td>
<td>A340-600</td>
<td></td>
<td></td>
<td>New Premium Economy Class for A340-600, the main cabin of the main floor on the aircraft, the change consists of the removal of 10% per cent of the Economy Class. Seats instead. The new seating will have a greater pitch.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- Supplemental Type Definition Document STDOS5-1343M Issue 1 rev 1, ICA, ref A0-08-1451-ICA, Issue 1 rev 0; Compliance Record List CRILS-13-42M Issue 1 rev 0 or later revisions of the above listed documents approved by EASA.</td>
<td>According to the design related to this STC, this modification must be reinstated in conjunction with a) SB AIM A340-25-107, 2013, Introduction of a three part class dividers installation (Minor Change of MM 636-14-1)</td>
<td>13/11/2014</td>
<td>Active</td>
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<td>10051172</td>
<td>LEAT S.p.A.</td>
<td>F27 MARK 500</td>
<td></td>
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<td>HF COM system installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Data List 23-10-0513 REV 00 dated 30 October 2014 Compliance Check List 23-10-0513-1100 REV 00 dated 30 October 2014 Design Office Substantiation 23-10-0513-1400 REV 00 dated 30 October 2014 Aircraft Flight Manual Supplement 23-10-0513-0800 REV 00 dated 30 October 2014 Instructions for Continued Airworthiness</td>
<td>None</td>
<td>14/11/2014</td>
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<td>10051176</td>
<td>ALTRAN LAB/ AIRCAD</td>
<td>A320-232</td>
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<td>Dragonair Cabin Reconfiguration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Data List MDS 13021 Rev 3, dated 12/11/2014 Compliance Check List CCL 13021 Rev 3, dated 29/10/2014 Instructions for Continued Airworthiness ICA 13021 Rev 0, dated 29/10/2014 or later revisions of the above listed documents</td>
<td>This STC does not constitute an approval for the installation of the Panasonic ExW IFE system, related wirings and its activation. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introd...</td>
<td>14/11/2014</td>
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<td>10051177</td>
<td>MARSHALL AVIATION SERVICES LIMITED</td>
<td>B200</td>
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<td>Installation of Transponder Enhanced Surveillance Installation of Transponder Enhanced Surveillance for Rockwell Collins TDFR-940 equipped King Air B200 Prolin II accordance with Marshall Aviation Services Ltd document SC0473B</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Installation of Transponder Enhanced Surveillance accordance with Marshall Aviation Services Ltd document SC0473B and Marshall Aviation Services Ltd document SC0473B and Marshall Aviation Services Ltd document SC0473B and Marshall Aviation Services Ltd document SC0473B</td>
<td>None</td>
<td>14/11/2014</td>
<td>Active</td>
<td>10051177</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10051180</td>
<td>ALITALIA SOCIETÀ AEREA ITALIANA S.p.A.</td>
<td>REV. 1</td>
<td></td>
<td>A320-214, A320-216, A319-111, A319-112</td>
<td>Installation of ADBSAFE passenger seat cargo container (P/NAC80-02207-81) Rev. 1. Transfer of Certificate</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-0334-01 Installation of cargo seat bags. The following paragraph(s) at a later amendment: CS-25, Amendment 4. For the paragraphs affected by the change...</td>
<td>Master Data List MM_MDL14F25S-0001 R02, dated 24/10/2014</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2021</td>
<td>Active</td>
<td>10051180</td>
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<tr>
<td>10051181</td>
<td>PW ENGINEERING GmbH</td>
<td>REV. 1</td>
<td></td>
<td>A380-841, A380-842, A380-851</td>
<td>PW - Structural Provisions of Patient Transport Compartment (PTC) on Airbus A380</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Instructions for Continued Airworthiness, PFW reference A388A-ICA-AS Issue Sep 15/2014; Master Drawing List, PFW reference A388B-MDR-53010 Issue C-1; Master Certification Documentation List, PFW reference CCN-A388A-MCDL Issue 1-Operators Manual, PFW reference A388A-COM-A Issue Sep 15/2014 or later revisions of the above documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/11/2014</td>
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<td>10051181</td>
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<td>10051182</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>REV. 1</td>
<td></td>
<td>AS 350 B2, A3 350 B3</td>
<td>AS 350 B2, A3 350 B3</td>
<td>Long-Line Electrical Provisions P/N 350-900594 in accordance with MDL-12-008 to provide electrical power for optional equipment.</td>
<td>Flight Manual Supplement FMS-HCA-152, Revision 1 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement Compliance Program CP-0369 and Balance Analysis R01, dated 22/09/2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/11/2014</td>
<td>Active</td>
<td>10051182</td>
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<td>10051201</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 1</td>
<td></td>
<td>737-600/-700/-800/-900-900ER</td>
<td>Retrofit Installation of FDAU SAGEM CRI D-0334-01 Installation of cargo seat bags. The following paragraph(s) at a later amendment: CS-25, Amendment 4. For the paragraphs affected by the change...</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-0334-01 Installation of cargo seat bags. The following paragraph(s) at a later amendment: CS-25, Amendment 4. For the paragraphs affected by the change...</td>
<td>Master Data List MM_MDL14F25S-0001 R02, dated 24/10/2014</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/11/2014</td>
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<td>10051207</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>REV. 1</td>
<td></td>
<td>35A, 36, 36A, 3S</td>
<td>TCAS TP187A to B Upgrade to Version 7.1 (Honeywell) Revision 1, replaces initial issue of this approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement, AG878-01, rev RInstructions for Continued Airworthiness, AG878-050, 8R, dated 22 Sep 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>18/11/2014</td>
<td>Active</td>
<td>10051207</td>
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List of EASA Supplemental Type Certificates

10051211

SOGECLAR AEROSPACE LIMITED

STC Holder: CASA UK RESPONSIBILITY

STC No.: 10051268

WIP Cabin Completion as defined in the:

Associated Technical ROEDER

Date of Active Limitations/Conditions

19/11/2014 Active

10051283

A320-214

737-700 IGW

REV. 4

Installation and activation of an In-Seat Power System on existing business class seats.

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.

STC valid for MSN 38408. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

STC holder: Ogden Aerospace


17/11/2014 Active 10051211

10051266

DELTA ENGINEERING CORPORATION

FAA STC ST0189NY

737-700 737-800

Installation of Alternative Wheel and Brakes

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25 change 4: 25.883; 1369; 1363; 143/1OS 25 Change 13: 25.831; 853; 881; 1563; 1585. The requirements for environmental protection and the associated certificated noise...


None 19/11/2014 Active 10051266

10051267

DELTA ENGINEERING CORPORATION

A330-323/-341/-342/-343

A330-301/-302/-303/-321/-322

A330-243/-243F


None 19/11/2014 Active 10051267

10051268

ROEDER PRAEGEN GMBH

REV. 1

A320-214

Installation and activation of an In-Seat Power System on existing business class seats.

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

STC holder: Ogden Aerospace


15/06/2018 Active 10051268

10031283

JET AVIATION AG

REV. 4

WIP Cabin Completion as defined in the:

- A330-301/-341/-342/-343, A330-323/-341/-342/-343 for EASA and TSO A

STC holder: Ogden Aerospace


17/10/2017 Active 10031283

An agency of the European Union
<table>
<thead>
<tr>
<th>STC Holder</th>
<th>FAA STC No</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>FOKKER SERVICES B.V.</td>
<td>10051300</td>
<td>Installation of TCAS to V7.1 on Boeing 737 Revision 1: Addition of introduction of TCAS V7.1 upgrade on TCAS Equipment Revision 2: Addition of introduction of TCAS V7.1 upgrade on AC/STCA22000 system for 737NG</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Original STC Certification Plan CP-0237 Issue 1Compliance Record Report CR-0237 Issue 3Compliance Record Report CP-0238 Issue 1Revision 1: In addition to the above documents Certification Plan CP-0260 Issue 2Compliance Record Report CR-0260 Issue 2Revision 2: MCA-22000 Installation of EASA EASA ED Decision 2004/04/01 etc. (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/EUSA Bilateral Agreement. Prior to installation of this design change it must be determined that the interchangeability between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/08/2015</td>
<td>Active</td>
<td>10051300</td>
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<td>DASSAULT FALCON JET CORPORATION</td>
<td>10051323</td>
<td>Validation of FAA STC Installation of LED Landing Light</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List ref: 1024415 rev. C, dated 13 February 2011 for later revisions of the above listed documents approved by EASA in accordance with EASA EASA ED Decision 2004/04/01 etc. (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/EUSA Bilateral Agreement.</td>
<td>24/11/2014</td>
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<td>Foreign/NAA Code</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>REV 1</td>
<td>FAA STC ST03252CH</td>
<td>MYSTERE FALCON 900 MYSTERE FALCON 50 FALCON 900EX</td>
<td>Validation of FAA STC Installation of LED Landing Lights</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Master Data List 1024638 rev. B, dated 18 February 2014 for later revisions of the above listed documents approved by EASA in accordance with EASA Ed Decision 2004/04/CF (or subsequent revisions off this decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this change and any other previously installed design change will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/12/2014</td>
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<tr>
<td>PMV ENGINEERING</td>
<td></td>
<td>737-700</td>
<td></td>
<td>eConnect IFE installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements D-118 Glass monitors issue 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Master Data List SCA-00340-MDL Rev 06/AFMS SCA-00340-AFMRev 03/AFMS SCA-00340-AMMS Rev 06/ for later revisions of the above listed documents approved by EASA.</td>
<td>The use of transmitting portable electronic devices (T-PEDs) - is allowed during non critical flight phase in the cabin but limited to WiFi devices - is NOT allowed during critical flight phase in the cabin - is NOT allowed during any flight phase in the cockpit. Prior to installation of this design change.</td>
<td>24/11/2014</td>
</tr>
<tr>
<td>AIRCRAFT STRUCTURES INTERNATIONAL CORP.</td>
<td>FAA STC SA01747W</td>
<td>208B 208</td>
<td>Installation of Pilot and Co-Pilot Lower Door Panels</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>ASIC Installation Data List 09251001-002 Rev. 1, dated 6 January 2014 for later revisions of the above listed documents approved by EASA in accordance with EASA Ed Decision 2004/04/CF (or subsequent revisions off this decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement.</td>
<td>If the holder agrees to permit another person to use this certificate/holder the product, the holder shall give the other person written authorization of that permission. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other pre...</td>
<td>24/11/2014</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>A330-200 (A330-243)</td>
<td>Thomas Cook - Cabin rearrangement / IFE installation/ Cabin seating rearrangement (&quot;A30 Y/C&quot; and &quot;ZIM&quot; Y/C Premium new seats)/ Zodiac &quot;1ra RAVE&quot; and new IFE system</td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- Change Approval Sheet CAS-AS15-14-002 rev. 0, dated 24 November 2014 - Instruction for Continued Airworthiness, ref: 1CA A330-14-002 rev. 0 dated 21 July 2014 for later revisions of the above listed documents approved by EASA.</td>
<td>- This STC is applicable to MSN 254, 266, 301 &amp; 309 only - Refer to the section &quot;B- Concurrent Requirements&quot; of the AIR FRANCE SERVICE BuLETIN reference AF38- A330-14-002-25-001. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other pre...</td>
<td>25/11/2014</td>
<td>Active</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST04296AT-D</td>
<td>GV</td>
<td>Installation of Mode AWK-6232 Series dual radio wireless access point.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Index List CE51 4286808 rev - Weight and Balance statement CE51 4289005 rev All Instructions for Continued Airworthiness CE51 4289011 rev - AFM Supplement CE51 4289011 rev - AFM Supplement CE51 2004/04/CF (or subsequent revisions).</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other pre...</td>
<td>26/11/2014</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>Gülfstream Aerospace Corporation</td>
<td>G-SV</td>
<td>Installation of Mode A/WK-632 series dual radio wireless access point.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Index (CE) 42839001 rev AIWeight and Balance statement CE4 42839001 rev Instructions for Continued Airworthiness CE4 42839007 rev AJM Supplement CE4 42838504 for Aircraft revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/4/CF (or subsequent revs)</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a noticeable effect upon the airworthiness of the product.</td>
<td>26/11/2014</td>
<td>Active</td>
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<td>Tenencia Ltd.</td>
<td>AE21-272N, AE21-271NX</td>
<td>Wireless IFE System Installation/Revision 2: Addition of new models</td>
<td>The Certification Basis for the original product as amended by the following additional alternative airworthiness requirements: Special Condition (s): Special Condition H-01 Enhanced Airworthiness Programme for Aeroplane System – ICAO on ESWI: The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Report TEN 751-47 Airplane Flight Manual Supplement, Airbus A330 Series/Wireless IFE System, Issue 09 Jan 2018/Tenencia Ltd. report Modification TEN 751, Airbus A330 Series Wireless IFE System Installation, Issue 1 dated 21 Nov 2014 and all referenced documents/later revisions of the above listed documents</td>
<td>Operational approval for the use of transmitting PEDs on board the aircraft must be obtained from the responsible division of the national authority of the aircraft's country of registration prior to conducting this kind of operation. Prior installation of this change/repair must be determined.</td>
<td>20/03/2018</td>
<td>Active</td>
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<td>Boggia Aeronautics S.R.L.</td>
<td>AB206 B, AB206 C</td>
<td>Fatigue Testing for AW limit extension/Rev. 1 extends lifetime of the kit</td>
<td>FAR 27 amendment 1: to 35 for the areas affected by the change and JAR27.856.2 change 2: The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise...</td>
<td>- The following documents supersede the associated Technical Documentation listed in the STC 10051391 dated 3 December 2014: - Airworthiness Limitations Section: 206P00210-ICA-00 - Chapter 04 - Issue 0 dated 7 May 2024 - Master Document List: 206P00208-ML-02 Issue 2 dated 3 June 2024 - Instructions for Cont...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce a noticeable effect upon the airworthiness of the product.</td>
<td>05/04/2024</td>
<td>Active</td>
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<td>Cesna Zurich Citation Service CentGmbH</td>
<td>CE210N 525, 520A, 522B</td>
<td>Compliance to ADS-B Out was shown against EASA CS ACN52ADS Band FAACAC20-165A as per RTCA D2068B2007 Implementation of ADS-B out capability into Rockwell Collins Pro Line 21-equipped aircraft, single or dual Rockwell Collins GPS54000e dual Rockwell Collins TD9494 Transponder single or dual FMS-3000R.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Instructions for continued airworthiness - ICA-111 Rev. C Airplane Flight Manual Supplement FMS-111 Rev. Cor later revisions of the above listed documents approved by EASA - Master Document List: MDL-111 Rev. B</td>
<td>Applicable Serial Numbers: CE210 C1+... SN 0600 and on CS215 CA22... SN 0300 and on CS22B C2J... SN 0001 thru SN40400 Pro Line 21 Aeronica Suite must be factory installed certified including MFS-3000 (single or dual) and single or dual GPS5400s (EGNOS/WAAS “1BS4” capable), dual Diversity...</td>
<td>11/12/2014</td>
<td>Active</td>
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<tr>
<td>Rodger Praesigion</td>
<td>P.68 “VICTOR” P.68R “VICTOR” P.68C-TC P.68B “VICTOR” P.68BTP “VICTOR” A6260P “SPARTACUS”</td>
<td>Installation of PFD Aspen EFO1000 PRO and MFD Avionix E368000th Sailton/Transceiver MLX770 and TAS805</td>
<td>Supplement to AFM P68- NY-04249-08 Rev. 00...; or later EASA approved revisions/Compliance Program P68- NY-04249-01 Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01a Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01b Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01c Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01d Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01e Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01f Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01g Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01h Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01i Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01j Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01k Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01l Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01m Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01n Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01o Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01p Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01q Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01r Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01s Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01t Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01u Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01v Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01w Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01x Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01y Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01z Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01aa Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01ab Rev 00 or later approved revisions/Compliance Checklist P68- NY-04249-01ac Rev 00...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a noticeable effect upon the airworthiness of the product.</td>
<td>15/11/2017</td>
<td>Active</td>
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<td>EASA STC No</td>
<td>STC Holder</td>
<td>Rev. level</td>
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<td>10051428</td>
<td>SIFARAN ELECTRONICS &amp; DEFENSE BUSINES SS UNIT - HELICOPTERS DATAMANAGEMENT SERVICES &amp; SERVICES</td>
<td>REV. 2</td>
<td>-</td>
<td>EC 155 B, EC 155 B1</td>
<td>Flight Data Monitoring and Transmitting System HELICOM V2 NUM / V2+ NUM</td>
<td>Installation Revision 1: Introduction of the HELICOM V2+ NUM version</td>
<td>- Revision 5 dated 14/01/2016: Correction Declaration - NW/EC 155-HV2+ NUM E-C-01, dated 14/01/2016</td>
<td>The approval holder shall fulfil the requirements for environmental protection and the associated certified noise and/or emission...</td>
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<td>10051429</td>
<td>S-TEC CORPORATION</td>
<td>FAA SA09536AC-D</td>
<td>PA-31T</td>
<td>Installation of S-TEC 01304 Digital Flight Control System Part N.01304-07-01-006</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- To be installed according to S-TEC Master Drawing List N ST-917-L-0001 rev H dated 02-Sep-2014; To be operated in accordance with S-TEC ARM-S PIN ST-917-4-001 rev C dated 04-Aug-2014; To be maintained i.a.w. Instruction for continued Airworthiness ST-917-C-0001 rev B dated 02-Aug-2005 or later.</td>
<td>- Limitations as per in original FAA STC SA09536AC-D dated 03-Sep-2014 - Installer must check availability of EASA approval for any referenced STC. Prior to installation of this design change it must be determined that the installation is compatible with the original product.</td>
<td>02/12/2014</td>
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<td>10051435</td>
<td>FR AVIATION Ltd/DIB COBHAM AVIONICS SERVICES UK</td>
<td>CAA UK RESPONSIBILITY</td>
<td>DORNIER 228-202</td>
<td>ACAS II Software Version 7.1 upgrade.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Cobham Modification DC-0252. Operation in accordance with Cobham Flight Manual Supplement refRANM/0125 Rev 0 dated November 2014. Maintain in accordance with Cobham ICA ref ICA-0252 Issue A dated 17 November 2014 or later revisions of the above listed documents approval.</td>
<td>Prior to installation of this design change it must be determined that the installation is compatible with the original product.</td>
<td>02/12/2014</td>
</tr>
<tr>
<td>10051448</td>
<td>FOKKER SERVICES B.V.</td>
<td>REV. 1</td>
<td>-</td>
<td>DHC-8-401/DHC-8-402 DHC-8-315 DHC-8-311/DHC-8-314 DHC-8-201/DHC-8-202 DHC-8-102/DHC-8-103/ DHC-8-106</td>
<td>Upgrade of TCAS 7.0 to 7.1 on DHC-8 100/200/300/400 Series</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation must be in accordance with Fokker Engineering BulletinEBDH-3-0013 Operation in accordance with AFM Supplement DHC-8-810 AM001 Rev 0 dated 17 November 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the installation is compatible with the original product.</td>
</tr>
<tr>
<td>10051464</td>
<td>CESSNA ZURICH CITATION SERVICE CENTGmbH</td>
<td>560 (VARIANT ENCORE *)</td>
<td>-</td>
<td>Compliance to ADS-B Out was shown against EASA CS ACN-352 Amd 1 FAAAC2016A as per RTCA DO289s Implementation of ADS-B out capability into Rockwell Collins Pro Line 21 equippied aircraft with an ICAO lumber.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Instructions for continued airworthiness - ICA-113, Rev. A - Aircraft Flight Manual Supplement - FMS-113, Rev. A prior to installation of this design change it must be determined that the installation is compatible with the original product.</td>
<td>Applicable Serial Numbers: C560 Encore - SN 560-0751 and on C560 Encore Pro Line 21 Avionics Suite must be factory installed in accordance with the above listed documents approved by EASA.</td>
<td>03/12/2014</td>
</tr>
</tbody>
</table>
## List of EASA Supplemental Type Certificates

<table>
<thead>
<tr>
<th>STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<th>System Record</th>
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</thead>
<tbody>
<tr>
<td>10051474</td>
<td>CESSNA-ZURICH CITATION SERVICE CENTGmbH</td>
<td>B200GT, B300, B300C</td>
<td>Medical</td>
<td>Compliance to ADS-B Out was shown against EASA CS ACSN, ADBand FAAAC0-15A as per RTCA DO220A/Implementation of ADS-B out capability into Rockwell Collins Pro Line 21-equipped aircraft with - single or dual Rockwell Collins GPS6000S-dual Rockwell Collins TDR94D Transponders.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this Certificate.</td>
<td>- Instructions for continued airworthiness - ICA-109, Rev R: Airplane Flight Manual Supplement - FMS-109, Rev R: Air later revisions of the above listed documents approved by EASA - Master Document List - MLI-109, Rev B</td>
<td>Applicable Serial Numbers: All except B200 serials BB-1385, BB-1386, BB-1387, BB-1388 and (B200C) BV11 and BV12 (UC-12Moah military designation RC-12M), BU-11 and BU-12 (UF-12F with military designation RC-12F) as per TCD94-A.277B(0)0X(03)0 (X). Pro Line 21 Antonia Suite must be factory installed...</td>
<td>04/12/2014</td>
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<td>10051474</td>
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<td>10051480</td>
<td>E/ Aerospace, Inc. - FSI</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>747-400</td>
<td>Deutsche Lufthansa 747-400 Airworthiness Limitations Mobiles (ALM) (s) Update - This major change updates the ALM(s) listed in the�프로젝트 and damage tolerance analysis and instructions for continued airworthiness. There is no design change.</td>
<td>The Certification Basis for the original product remains applicable to this certificate.</td>
<td>TF-P12-218AFM - Flight Manual Supplement for Lufthansa Airlines (2014 SEP724)-B2FS218-D15 Chapter 10 - Airworthiness Limitations (Rev B, 2014 NOV 11)--B2FS218-D15 - Supplement To Instructions For Continued AirworthinessFor DLH- BT747-400 UpperDeck Interior Reconfiguration and Door C7 Steer Neal revC, 29...</td>
<td>06/11/2015</td>
<td>Active</td>
<td>10051480</td>
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<tr>
<td>10051487</td>
<td>Dassault Falcon Jet Corporation</td>
<td>FAA STC</td>
<td>ST90302SC-D</td>
<td>FAULCON 2000</td>
<td>Validation of FAA STC ST90302SC-D Installation of Speed Indextor System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Master Drawing List ref. F2ML10-30 rev. D, dated 27 July 2000 Airplane Flight Manual Supplement ref. 36528-1 rev. A, dated 14 June2000 or later revisions of the above listed documents approved by EASA in accordance with EASA Ed Decision 2004/04/CF (or subsequent revisions off this decision) and or the...</td>
<td>06/12/2014</td>
<td>Active</td>
<td>10051487</td>
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<tr>
<td>10051489</td>
<td>Dassault Aircraft Services (DAS)</td>
<td>FAA STC</td>
<td>ST90657NY</td>
<td>MYSTERE-FALCON 900</td>
<td>Validation of FAA STC Installation of Honeywell GPS/Adaas System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>STC Master Data List F2BN-00012-006 rev. D, dated 10 October 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA Ed Decision 2004/04/CF (or subsequent revisions off this decision) and or the Technical Implementation Procedures of EUUSIA Bilateral Agreement...</td>
<td>For MF900 model, applicability limited to P900C version Prior to installation of this design change it must be determined that the...</td>
<td>05/12/2014</td>
<td>Active</td>
<td>10051489</td>
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<tr>
<td>10051493</td>
<td>Dassault Aircraft Services (DAS)</td>
<td>FAA STC</td>
<td>ST90979NY</td>
<td>MYSTERE-FALCON 900</td>
<td>Validation of FAA STC Installation of Honeywell GPS/Adaas System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>STC Master Data List F2BN-00024-006 rev. A, dated 10 December 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA Ed Decision 2004/04/CF (or subsequent revisions off this decision) and or the Technical Implementation Procedures of EUUSIA Bilateral Agreement...</td>
<td>Applicable to MF900 model, except P900C version Prior to installation of this design change it must be determined that the...</td>
<td>05/12/2014</td>
<td>Active</td>
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<td>STC Holder</td>
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<tr>
<td>AVIONICS DESIGN ApS</td>
<td>Major Change to STC for TCAS 7.1 UpgradeRevision 1 - Honeywell TCAS computer changed part number from 90-0581-001 to 940-0581-005 to correct False RA issue</td>
<td>Engineering Order &amp; Certification Program for TCAS 7.1 (ACAS II) Upgrade Prior Existing TCAS 7.0 in Airbus A318, A319, A320, A321: Documento0285803-EO Issue 04 dated 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Master Data List ref. AED2014-27M MDL. rev. A0 dated 05 December 2014.AFM Supplement ref. AED2014-27M AFMS rev. A0 dated 04 December 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/10/2017</td>
<td>Active</td>
<td>10051519</td>
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<td>SABENA TECHNICS BGC</td>
<td>Y/C Cabin Modification with Zodiac Inflight Innovations (Zii) Rave IFE Rave installation / upgrade on A330-200 this STC installs the Rave IFE system, which mainly consists in the integration through the current cabin arrangement of System Control Unit, Seat power unit, Seat Display Unit, Crew Panel and...</td>
<td>Engineering Order &amp; Certification Program for TCAS 7.1 (ACAS II) Upgrade Prior Existing TCAS 7.0 in Airbus A318, A319, A320, A321: Documento0285803-EO Issue 04 dated 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Master Data List ref. 4833-01A1-2103- R00Instruction for Continued Airworthiness, ref : 4833-01A-2211- R01AirCraft Flight Manual Supplement, ref : 4833-01A-2501-900or later revisions of the above listed documents approved by EASA.</td>
<td>1) This STC is applicable only to MSN 303, MSN 306, MSN 311-2 EASA STC ref 10037643 (Initial issue, or later approved revision).Installation of new Business Class Avio EBB2 Seatand IMS Rave IFE System: must be concurrently embodied.</td>
<td>09/12/2014</td>
<td>Active</td>
<td>10051534</td>
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<td>SABENA TECHNICS BOD</td>
<td>Installation...</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List ref. 4833-01A1-2103- R00Instruction for Continued Airworthiness, ref : 4833-01A-2211- R01AirCraft Flight Manual Supplement, ref : 4833-01A-2501-900or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/12/2014</td>
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<td>10051559</td>
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<td>STC STC ref.</td>
<td>Rev. level</td>
<td>Foreign/NAAA STC ref.</td>
<td>Models</td>
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<td>Associated Technical Documentation</td>
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<td>10051593</td>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>REV. 1</td>
<td>727-200</td>
<td>CPDLC Installation Rev. 1: Transfer of ST to FMS</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25 Amendment 1426, 15101, 25.1302, 25.1305, 25.1306, 25.1309, 25.1310, 25.1312, 25.1315, 25.1353, 25.1355, 25.1356, 25.1357, 25.1359, 25.1360, 25.1361, 25.1362, 25.1363, 25.1364, 25.1365, 25.1366, 25.1367, 25.1368, 25.1369, 25.1370.</td>
<td>Flight Manual Supplement 12587-APM-0001 Issue 3 for later revisions of the above listed document(s) approved/accepted under the EASA system Master Data List MDL-12587-0001 Issue 3</td>
<td>Prior to installation of the change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/11/2021</td>
<td>Active</td>
<td>10051593</td>
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<tr>
<td>10051595</td>
<td>AIRGLAS, Inc.</td>
<td>FAA SR02430AK</td>
<td>R 44, R 44 II</td>
<td>Installation of LRSCE-R44 Rear Seat Cargo Extension</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Retrorocket Flight Manual Supplement (RFMS) for Robinson models R44 and R44 II equipped with Airglas LRSCE-R44 Rear Cargo Seat Extension (rev. a) for LRSCE-R44-AR00 and 07-4280 (multi block installation).</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/12/2014</td>
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<td>10051595</td>
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<td>10051609</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>BOEING 747-400</td>
<td>Lufthansa 8747-4 K Retr: This Modification reflects an interchange of all Business Class Seats, which results in an increase of 59 passengers. A new IFE System will be incorporated, which also involved a modification of all Economyseats, Partitions and Monuments.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CRI D-101 - Width of aisle - 4 passenger configuration (closed at issuing on 22 July 2014). The following paragraph(s) at a later amendment: CRI 25-00-01/Annex 14 for later revisions of the above listed documents approved by EASA.</td>
<td>This modification must be installed in conjunction with Flightstructures inc. EASA Supplemental Type Certificate 10055480.</td>
<td>12/12/2014</td>
<td>Active</td>
<td>10051609</td>
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<td>10051622</td>
<td>ALOFT AEROARCHITECT S</td>
<td>REV. 5</td>
<td>FAA ST01716NY</td>
<td>737-700</td>
<td>Installation of Auxiliary Fuel Systems in Boeing 737-700/800(WL)(J) Compliant to § 25.981 amendment 25-125 - All configurations listed below are approved with or without Boeing Aircraft Generation System (NGS): installed - Configuration 1-E: Installation of a Two-Call (One Forward, One Aft) Auxiliary Fuel.</td>
<td>The Certification Basis for the original product remains applicable to all certificates/approval, except where amended by additional or later amendments: Special Conditions in CRI E-101 &quot;Flammability reduction&quot;. The requirements for environmental protection and the associated certification noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List No. 5888-280-001, Rev. AB dated 28 Oct 2020: 1. Configuration 1-E: Installation of a Two-Call (One Forward, One Aft) Auxiliary Fuel System in the forward and aft cargo compartment in accordance with Master Drawing List No. 5888-280-001, Rev. G dated 04 November 2011, 2A -22LM System...</td>
<td>The aircraft must be inspected in accordance with the requirements in the Airworthiness Limitations. Section 4 and 4a in the MPDS for other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/06/2021</td>
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<td>STC Holder</td>
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<tr>
<td>DSA A.S.</td>
<td>PA-31-350 (CHIEFTAN) PA31-325 PA31-200 PA31</td>
<td>Installation of an IGPS (Identification of Gross Polluting Ships) monitoring system.</td>
<td>- COMPLIANCE CHECKLIST 02/03/21536-001, Doc. LDA 02/13/PA31, rev. 3.1 dated 13.11.2014.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/05/2020</td>
<td>Active</td>
<td>10051623</td>
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<tr>
<td>PAUSTIAN ARTEX GmbH</td>
<td>B737-800</td>
<td>Installation of placards on the aircraft cabin windows inner panel.</td>
<td>Master Documentation List MDP-016 rev. 0 (Instructions for continued airworthiness ICA-DP158 rev. 0)</td>
<td>Pending positive results of the Entry into Service phase tests (as per document TP-DP16-2, this certificate is limited to the entry into service test phase of 6 months after issuance of this certificate) the aircraft MSN 25350. Prior to installation of this design change it must be determined that the change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/12/2014</td>
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<tr>
<td>SPECTRUM AEROMED</td>
<td>AW139, AB139</td>
<td>Air Ambulance Conversion EASA STC 10051639 Rev. 1 reflects the transfer of EASA STC 10051639 from SNAKY to SPECTRUM AEROMED.</td>
<td>Master Documentation List MDP-016 rev. 0 Instructions for continued airworthiness ICA-DP158 rev. 0</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/04/2019</td>
<td>Active</td>
<td>10051639</td>
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<tr>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>EC 130 T2</td>
<td>Cargo Mirror Installation. The installation consists of two mirrors mounted externally, underneath the cabin in front of the pilot, on a support frame which is feed to the helicopter four attachment points on the left hand side of the aircraft.</td>
<td>Airbus Helicopters Canada Limited Compliance Program. TTP document TP14-MDL-001. 2014-06-05.</td>
<td>None.</td>
<td>15/12/2014</td>
<td>Active</td>
<td>10051642</td>
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<tr>
<td>BUCHER LEICHTBAU AG</td>
<td>MBB-BK117 D-34 MBB-BK117 D-3 MBB-BK117 D-2 MBB-BK117 D-2 Project Certification Plan E026-05-15 Rev. 3.</td>
<td>Emergency Medical System (EMS) P/N ARA-EC145-AC7 Rev. 1 follow on certification of the oxygen system as described in Certification Plan E026-54-14 Rev. 2 follow on certification of the additional Oxygen Bottle as described in Certification Plan E026-05-15 Rev. 3</td>
<td>Flight Manual Supplement ARA-EC145-AC70 Rev. 2 (labelled as Continued Airworthiness ARRA-EC145-AC70 Rev. 2)</td>
<td>For Rev. 3: Subject change is only applicable to MBB-BK117 D-3 stroker having serial number 20050 and above or having serial numbers below 20060 with incorporated SB MBB-BK117 D-2-SB-005-01, or later approved revisions. Prior to installation of this change/repair it must be determined that the change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/11/2021</td>
<td>Active</td>
<td>10051657</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>PMV ENGINEERING</td>
<td>A340-643, A340-641, A340-52, A340-642, A340-311, A340-312, A340-313, A340-211, A340-212, A340-213</td>
<td>TCAS 7.1 upgrade on Airbus A340. The change consists of the upgrade of the current TCAS II software version 7.0 to TCAS II software version 7.1 on A340 family aircraft and its update to EASA STC ref. 10051669.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- MASTER DOCUMENT LIST, ref: AV-01710-MDL- R06 dated 17/11/2017; - INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, ref: AV-00710-IAM-R01 dated 28/05/2017; - AIRCRAFT FLIGHT MANUAL SUPPLEMENT, ref: AV-00710-APS-R03 dated 03/07/2017 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/11/2017</td>
<td>Active</td>
<td>10051669</td>
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<tr>
<td>GFD GmbH</td>
<td>31A/35A/36A, 31A, 35A, 36A</td>
<td>Installation of S-TEC System 55/55X Two PMV</td>
<td>AFM Supplement Doc. No. S254-071 Rev. IR of 10.12.2014 or later revisions of the above listed documents approved by EASA Master Data List Doc. No. S254-049 Rev. B of 10.12.2014 Instructions for Continued Airworthiness Doc. No. S254-16 Rev. A of 11.08.2014 or later revisions of the above listed documents.</td>
<td>Limited to GFD Fleet. D-24806 Hohen, Germany. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
<td>Active</td>
<td>10051674</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA0911AC-D</td>
<td>Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, (Model ST-550)</td>
<td>Installation, according to FAA STC SA0911AC-D, Revision 4, dated 03-02 and Master Drawing List No. 92792, Revision D, dated 03-02 (14 Volt System) Operations and Maintenance according to FAA/DAS Approved Pilot Operating Handbook and/or Airplane Flight Manual Supplement, PN 881177 or later revisions of...</td>
<td>1. Model M20D eligibility limited to aircraft modified per Mooney Drawing No. 950882 (Retractable Gear conversion). 2. FAA/DAS Approved Pilot Operating Handbook and/or Airplane Flight Manual Supplement, PN 881177, dated 11.08.14 or later revisions of...</td>
<td>18/12/2014</td>
<td>Active</td>
<td>10051678</td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>M20C, M20D, M20E</td>
<td>Installation of S-Blade MTV-9-B-C-I/18 on Extra 200</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation, according to FAA STC SA0911AC-D, Revision 4, dated 03-02 and Master Drawing List No. 92792, Revision D, dated 03-02 (14 Volt System) Operations and Maintenance according to FAA/DAS Approved Pilot Operating Handbook and/or Airplane Flight Manual Supplement, PN 881177 or later revisions of...</td>
<td>According to the previous installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2014</td>
<td>Active</td>
<td>10051680</td>
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<tr>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>A36</td>
<td>Installation of S-Blade MTV-5-B-C-D-C/F-R (A/131R10-56) on A38 Bonanza</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>According to the previous installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2014</td>
<td>Active</td>
<td>10051683</td>
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</table>
The Certification Basis for the original product remains applicable with certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. It remains applicable to this certification/approval without any impact.

The Certification Basis for the original product remains applicable with certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.

Installation of aft galley/lavatory complex.

The Certification Basis for the original product remains applicable with certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Installation of OnAir Inflight Connectivity System.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) of a later amendment. JAR 25 Change 13 for DHC-8, (a) and (c), 25.812 (a) and (e), 25.853 (a), (b) and (c). The requirements for environmental protection and the associated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.

Installation of Sagem Aircraft Pitting Inertial Reference Sensor (APRIS) R227 - AHSIS Upgrade.
<table>
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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Document</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10051708</td>
<td>AIRBUS DEFENCE AND SPACE S.A.</td>
<td>REV. 1</td>
<td>A320-231/230-233 A320-211/212-214/215/-216</td>
<td>Installation of VHF/UHF radio on Airbus Corporate Jet aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate approval.</td>
<td>Certification Plan MEA-DA-DT-140003 Issue D, dated 18/12/2014</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>19/12/2014</td>
<td>Active</td>
<td>10051708</td>
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<tr>
<td>10051710</td>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td></td>
<td>TCCA SA15-19 DHC-8-201-202 DHC-8-200-200-103-108</td>
<td>Triple Litter Stacking System Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate approval.</td>
<td>Certification Plan MEA-EngineeringOrder CE14-401, Issue C. Maintenance in accordance with Avmax Aviation Services Inc. Maintenance/Repair Manual Supplement CE14-701, Initial Issue Operation in accordance with Avmax Aviation Services Inc. Flight Manual/Supplement CE14-401.</td>
<td></td>
<td>The compatibility of stretchers to be installed must be approved by the responsible authority. Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect.</td>
<td>19/12/2014</td>
<td>Active</td>
<td>10051710</td>
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<td>10051729</td>
<td>AVIATION SPECIALTIES UNLIMITED, INC.</td>
<td>REV. 3</td>
<td>FAA SR01207SSE</td>
<td>EC135T3, EC135P3 EC131T1, EC131T2, EC131P2, EC131P1, EC131P2, EC131P4.</td>
<td>Installation of Night Vision Imaging System (NVIS) compatible lighting: Rev. 1: Installation of SRN 0033 to the MDL-135004-3Rev. 2: Addition of model EC135PRev. 3: Addition of model EC135T3 with Helinix cockpit</td>
<td>Certification Plan ASU-01207SSE</td>
<td></td>
<td></td>
<td>12/12/2019</td>
<td>Active</td>
<td>10051729</td>
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<tr>
<td>10051733</td>
<td>DELTA Q DESIGN Inc.</td>
<td></td>
<td>A321-213/-212/-220 A320-211/-212/-220 A320-212/-214/-220 A319-111-112/-113/-114/-115</td>
<td>Structural modification necessary for the installation of an IGA-5001 Intermediate Gain Antenna</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25.251(a)(b)(c) and 25.351 Amdt 0, 25.365(e) Amdt 0</td>
<td>Certification Plan Delta Q Design document No. DGD-IGA01-IC-01; Revision E dated 16 April 2014</td>
<td></td>
<td></td>
<td>Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>19/12/2014</td>
<td>Active</td>
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<td>10051734</td>
<td>AIRBUS S.A.S.</td>
<td></td>
<td>A319-115</td>
<td>Airbus Corporate Jet Modification C1700 VIP Project “Butterfly”</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s) CRI (E2-1: Isolated Compartments CRI-D-034-000 - Shower Installation CRI-D-035-004 - Minusius/Equivalent Safety Finding(s) CRI-E-28 - Improved Flamma...</td>
<td>Modification Approval Sheet C1700, Issue 1, dated 18/12/2014</td>
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<td>19/12/2014</td>
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### List of EASA Supplemental Type Certificates

<p>| STC Holder                     | Rev. level | Foreign/NAA STC ref. | Model(s)                                                                 | Description                                                                                               | Certification basis                                                                                       | Associated Technical Documentation                                                                 | Limitations/Conditions                                                                 | Date of Issue | Status | System Record |
|--------------------------------|------------|----------------------|--------------------------------------------------------------------------|----------------------------------------------------------------------------------------------------------|-----------------------------------------------------------------------------------------------------------|------------------------------------------------------------------------------------------------------------|------------------------------------------|-----------|----------------|
| STC Holder                     | Rev. level | Foreign/NAA STC ref. | Model(s)                                                                 | Description                                                                                               | Certification basis                                                                                       | Associated Technical Documentation                                                                 | Limitations/Conditions                                                                 | Date of Issue | Status | System Record |
| Lufthansa Technik AG            |            |                      | A320-231/-232/-233 A320-211/-212/-214/-215/-216                          | Installation of SWISS MAX cabin layout                                                                    | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.   | Supplemental Type Definition Document STC025-1343/M Issue 1Rev. 0 dated 03/12/2014 Compliance Record List CRL25-1343/M Issue 1 Rev. 1, dated 08/12/2014 for Continued Airworthiness A-00-14/53/ICA Ed 2 Rev. 0 dated 19/12/2014. | This modification must be installed in conjunction with NorthwindAerospace Technologies STC 10051689. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a adverse effect upon the airworthiness of the product. | 19/12/2014 | Active | 10051743     |
| EAD Aerospace Airworthiness     | REV. 2     |                      | A320-231/-232/-233 A320-211/-212/-214/-215/-216 A319-131/-132/-133 A320-111/-112/-113/-114/-115 | Activation for GIM/WFI Installation                                                                      | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.   | Master Drawing List, EAD Aerospace document No. 01442-EAD-A-MDL, Revision B, dated 18 December 2014. Instructions for Continued Airworthiness EAD Aerospace document No. 01442-EAD-A-ICA-01, Revision A, dated 26 September 2014. Flight Manual Supplement, EAD Aerospace document No.01442-EAD-A-AF. | prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a adverse effect upon the airworthiness of the product. | 07/01/2015 | Active | 10051755     |
| Société d'étude et de gestion aéronautique (SEGA) |            |                      | A36, A36TC                                                             | Increase of Maximum Take-off Weight (MTOW)                                                                | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.   | POH/AFM Supplement (Document No. PO-B36-57000) or later revisions of the above listed documents approved by EASA. | Provision of design change and any other previously installed design change and/or repair will not introduce a adverse effect upon the airworthiness of the product. | 22/12/2014 | Active | 10051760     |
| Aerostruktur Faserverbund Technik GmbH |            |                      | A36, A36TC                                                             | Increase of Maximum Take-off Weight (MTOW)                                                                | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.   | POH/AFM Supplement (Document No. PO-B36-57000) or later revisions of the above listed documents approved by EASA. | Provision of design change and any other previously installed design change and/or repair will not introduce a adverse effect upon the airworthiness of the product. | 22/12/2014 | Active | 10051763     |
| Aerotec &amp; Concept               |            |                      | 1908C                                                                  | Avionics update including GPS-460W, MDC-41 ACU, Sandel SN4500 and SN55500 in order to provide P-NAV &amp; LPV capability. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.   | Installation in accordance with AEROTEC-modification ref AE14230.Operate in accordance with AEROTEC Flight Manual Supplement refAE14230AFMS Rev 0 dated December 2014. Maintain in accordance with AEROTEC Instructions for Continued Airworthiness (ICA) ref AE14220ICA Rev 0 dated 20 October 2014. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a adverse effect upon the airworthiness of the product. | 22/12/2014 | Active | 10051769     |</p>
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<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>10051788</td>
<td>AEROMECH Inc.</td>
<td></td>
<td>727C 727-200F 727-700</td>
<td>727-100C 727</td>
<td>Group Approval for Reduced Vertical Separation Minimum</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement/727-03-10020 S Rev Cor later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/494/CF (or subsequent amendments or this decision) and/or the Technical Implementation Procedures of EUUSA Bilateral Agreement.Aircraft System...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise or smoke in a manner which is above the noise levels of the original product</td>
<td>23/12/2014</td>
<td>Active</td>
<td>10551788</td>
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<tr>
<td>10051820</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td></td>
<td>FAA STC ST01141CH</td>
<td>747-400F 747-400</td>
<td>Installation of a ROWmeet A specification electronic flight bag(ETF) class II-Configuration 1: Installation of Dual SDU to 747-200F and VB-11 P1 connection only-Configuration 2: Installation of Dual SDU to the original product as amendedby the following additional or alternative airworthiness requirements:CS25 (A)/(B): 25.29 (b), 25.301(3a), 25.305 (a) , 25.307(a), 25.307(c), 25.561 (c), 25.562(c)(6), 25.621, 25.603, 25.605 (a), 25.607, 25.609,25.611, 25.613(a), 25.615...</td>
<td>The Certification Basis for the original product as amendedby the following additional or alternative airworthiness requirements:CS25 (A)/(B): 25.29 (b), 25.301(3a), 25.305 (a) , 25.307(a), 25.307(c), 25.561 (c), 25.562(c)(6), 25.621, 25.603, 25.605 (a), 25.607, 25.609,25.611, 25.613(a), 25.615...</td>
<td>- Master Data List ECS-209959 Rev L- AFMS Configuration 1 ECS-209962 Rev B- AFMS Configuration 2 ECS-209935 Rev B- AFMSConfiguration3 ECS-209936 Rev B- AFMS Configuration 4 ECS-210440 Rev A- AFMS Configuration 5 ECS-Z10452 Rev A- Instructions for continued airworthiness ECS-209921 rev C- Weight and...</td>
<td>- EZ Mount mechanical mounting assemblies approved by EASA STC 10040901(FAA STC STA09070) is a prerequisite for the design changes described in this approval. Information displayed on the ETF is not to be used for primary navigation, and shall be used for reference only.- This STC does not provide o...</td>
<td>07/01/2015</td>
<td>Active</td>
<td>10551820</td>
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<tr>
<td>10051858</td>
<td>S-TEC CORPORATION</td>
<td></td>
<td>FAA SA09153AC-D</td>
<td>FR172 C, FR172 H, FR172 E, FR172 F</td>
<td>Installation of S-TEC System 75/55X Two Axis Automatic Flight Guidance System, Model ST-016 in accordance with FAA STC SA09153AC-D</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List 02025 Rev. C, dated June 1, 2019Installation Instructions Bulletin No. 716, Rev. 3, dated June 1, 2019Airplane Flight Manual Supplement 891257, Rev. 1, dated October 12, 2019or later versions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004...</td>
<td>If the holder agrees to permit another person to use this certificate to acquire the necessary parts and/or information to install the STC to another person, the holder must provide the person with written evidence of that permission. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair prohibits unrestricted flight operation.</td>
<td>09/01/2015</td>
<td>Active</td>
<td>10551858</td>
</tr>
<tr>
<td>10051880</td>
<td>CLUB AEROSTATIQUE DE FRANCHE-COMTE</td>
<td></td>
<td>GALAXY-7 &amp; GLX-1737</td>
<td>Installation of a Cameron Basket CB-300 installation on a Hot Air Balloon Galaxy 7 envelope, sn GLX-1737, with a Cameron Basket CB-300 installation on a Hot Air Balloon Galaxy 7 envelope</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Manual Supplement &lt;&lt;Supplement au manuel de vol V&gt;&gt; dated January 2015or later revisions of the above listeddocuments approved by EASA.</td>
<td>Modification limited to the Galaxy 7 envelope, sn GLX-1737, with a Cameron Basket CB-300 installation on a Hot Air Balloon Galaxy 7 envelope</td>
<td>08/01/2015</td>
<td>Active</td>
<td>10551880</td>
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<tr>
<td>10051871</td>
<td>ARLEIC AIR ESPACE</td>
<td></td>
<td>SA226-7 (B) SA226-7 SA226-AT</td>
<td>AAE004 - MVP-50T Display Engine</td>
<td>The Certification Basis for the original product as amendedby the following additional or alternative airworthiness requirements:the following paragraph(s) at a later amendment:Refer to certification program AAE004 CP document for paragraphs at CS23A/200 and 250.613(a), 250.615...</td>
<td>The Certification Basis for the original product as amendedby the following additional or alternative airworthiness requirements:the following paragraph(s) at a later amendment:Refer to certification program AAE004 CP document for paragraphs at CS23A/200 and 250.613(a), 250.615...</td>
<td>None</td>
<td>None</td>
<td>12/01/2015</td>
<td>Active</td>
<td>10551871</td>
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<tr>
<td>10051872</td>
<td>EMTEG CANADA LTD</td>
<td></td>
<td>TCCA C-LSA14-215D</td>
<td>CL-860-2D4</td>
<td>Removal of Optional FWD galley and lavatory</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Transport Canada, STC No. C-LSA14-215D issue 1 dated 05 December 2014EMTEG Canada Modification Data Summary J98300/Rev Csn/Roder later revisions of the above listed documents approved by EASA inaccordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement</td>
<td>Restricted to ASN 15065</td>
<td>12/01/2015</td>
<td>Active</td>
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<td>STC Holder</td>
<td>System</td>
<td>Active A340-311/-312/-313 CRANFIELD 10051890 DUNCAN 10051880 Certification Basis</td>
<td>Tune Box IFE and Satcom SBB</td>
<td>Rev. 1</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: C25.1301, C25.1309, C25.1681 - all at Amendment 4. The requirements for environmental protection and the associated... Installation must be in accordance with Technical Order Doc. 2000767-T. Issue 1. Operation must be in accordance with AFM Supplement Doc. 2000767-AM A Issue 1. Continued Airworthiness must be in accordance with ICA Doc. 2006767-CA Issue 1. Latest revisions of the above listed documents approved by EASA. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. Date of Issue: 12/02/2015 Status: Active System Record: 10051877</td>
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<td>System</td>
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<td>Tune Box IFE and Satcom SBB</td>
<td>Rev. 1</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: C25.1301, C25.1309, C25.1681 - all at Amendment 4. The requirements for environmental protection and the associated... Installation must be in accordance with Technical Order Doc. 2000767-T. Issue 1. Operation must be in accordance with AFM Supplement Doc. 2000767-AM A Issue 1. Continued Airworthiness must be in accordance with ICA Doc. 2006767-CA Issue 1. Latest revisions of the above listed documents approved by EASA. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. Date of Issue: 12/02/2015 Status: Active System Record: 10051877</td>
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<td>System</td>
<td>Active A340-311/-312/-313 CRANFIELD 10051890 DUNCAN 10051880 Certification Basis</td>
<td>Tune Box IFE and Satcom SBB</td>
<td>Rev. 1</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: C25.1301, C25.1309, C25.1681 - all at Amendment 4. The requirements for environmental protection and the associated... Installation must be in accordance with Technical Order Doc. 2000767-T. Issue 1. Operation must be in accordance with AFM Supplement Doc. 2000767-AM A Issue 1. Continued Airworthiness must be in accordance with ICA Doc. 2006767-CA Issue 1. Latest revisions of the above listed documents approved by EASA. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. Date of Issue: 12/02/2015 Status: Active System Record: 10051877</td>
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<td>System</td>
<td>Active A340-311/-312/-313 CRANFIELD 10051890 DUNCAN 10051880 Certification Basis</td>
<td>Tune Box IFE and Satcom SBB</td>
<td>Rev. 1</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: C25.1301, C25.1309, C25.1681 - all at Amendment 4. The requirements for environmental protection and the associated... Installation must be in accordance with Technical Order Doc. 2000767-T. Issue 1. Operation must be in accordance with AFM Supplement Doc. 2000767-AM A Issue 1. Continued Airworthiness must be in accordance with ICA Doc. 2006767-CA Issue 1. Latest revisions of the above listed documents approved by EASA. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. Date of Issue: 12/02/2015 Status: Active System Record: 10051877</td>
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<td>System</td>
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<td>Tune Box IFE and Satcom SBB</td>
<td>Rev. 1</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: C25.1301, C25.1309, C25.1681 - all at Amendment 4. The requirements for environmental protection and the associated... Installation must be in accordance with Technical Order Doc. 2000767-T. Issue 1. Operation must be in accordance with AFM Supplement Doc. 2000767-AM A Issue 1. Continued Airworthiness must be in accordance with ICA Doc. 2006767-CA Issue 1. Latest revisions of the above listed documents approved by EASA. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. Date of Issue: 12/02/2015 Status: Active System Record: 10051877</td>
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<td>STC Holder</td>
<td>System</td>
<td>Active A340-311/-312/-313 CRANFIELD 10051890 DUNCAN 10051880 Certification Basis</td>
<td>Tune Box IFE and Satcom SBB</td>
<td>Rev. 1</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: C25.1301, C25.1309, C25.1681 - all at Amendment 4. The requirements for environmental protection and the associated... Installation must be in accordance with Technical Order Doc. 2000767-T. Issue 1. Operation must be in accordance with AFM Supplement Doc. 2000767-AM A Issue 1. Continued Airworthiness must be in accordance with ICA Doc. 2006767-CA Issue 1. Latest revisions of the above listed documents approved by EASA. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. Date of Issue: 12/02/2015 Status: Active System Record: 10051877</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>FAA STC No.</td>
<td>EASA STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10051930</td>
<td>J.P. INSTRUMENTS, INC</td>
<td>FAA SAO20725E</td>
<td>SEE APPROVED MODEL LIST</td>
<td>10051937</td>
<td>Installation of twin temperature indicating system with fuel flow.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with J.P Instruments (JPI) Installation Manual Report No 760 Revision , dated 7/20/99. Operate in accordance with J.P Flight Manual Supplement ref No 760 dated August 31, 1999 or later revisions of the above listed documents approved by EASA.</td>
<td>Cylinder head, oil, turbine inlet and/or exhaust gas temperature, fuel flow equipment, tachometer instruments, and manifold pressure instruments required by the original type design, or if required, by their approval must be installed. Prior to installation of this design change it must be determined that...</td>
<td>16/01/2015</td>
<td>Active</td>
<td>10051930</td>
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<tr>
<td>10051935</td>
<td>J.P. INSTRUMENTS, INC</td>
<td>FAA STC SA01625SE</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>10051937</td>
<td>Installation of J P Instruments (JPI) EDM-960 Primary Engine DataManagement System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with J.P Installation Manual for EDM-960, Report No 909, Revision N0, dated February 22,2000. Operate in accordance with J.P FAA approved Flight Manual Supplement No960-001, dated March 27, 2000 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise or effect the performance of the FAA approved product.</td>
<td>19/01/2015</td>
<td>Active</td>
<td>10051935</td>
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<tr>
<td>10051937</td>
<td>J.P. INSTRUMENTS, INC</td>
<td>FAA SAO0815SE</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>10051941</td>
<td>Installation of J P Instruments (JPI) EDM-960 Primary Engine DataManagement System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with J.P Installation Manual FS-450, Report No 400, Revision - dated August 18, 2000. Operate in accordance with J.P Instruments Flight Manual Supplement ref EDM-960, dated December 18, 2000 or later revisions of the above listed documents approved by EASA.</td>
<td>This STC requires the installation of a fuel Flow Transducer per FAA STC/SAO4325SE, or aircraft listed on the FAA approved AML SAO04325SE and that have been previously modified with a fuel flow transducer P/N 201-A, 201-B, 201-C or 231 are eligible for installation for the FS-450, it is the installer responsibility...</td>
<td>19/01/2015</td>
<td>Active</td>
<td>10051937</td>
</tr>
<tr>
<td>10051939</td>
<td>J.P. INSTRUMENTS, INC</td>
<td>FAA SAO1436SE</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>10051939</td>
<td>Installation of J P Instruments (JPI) EDM-960 Primary Engine DataManagement System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with J.P Installation Manual for the EDM-960/930, Report 909, dated September 9, 2004. Operate in accordance with J.P Flight Manual Supplement ref EDM-900 and EDM-930, Primary Engine Data Management System, dated December 10,2004. Document Numbers 900-0001 and 900-0001 Rev...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise or effect the performance of the FAA approved product.</td>
<td>19/01/2015</td>
<td>Active</td>
<td>10051939</td>
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<td>10051947</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC</td>
<td>REV. 2</td>
<td>737-700</td>
<td>10051947</td>
<td>Interior Reconfiguration of KLM</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>-Master Drawing List, Northwest Aerospace Technologies, Inc. documentNo. N0485-D10, Revision B, dated 13 January 2015.-Certification Summary, Northwest Aerospace Technologies, Inc. documentNo. N0485- D90, Revision -, dated 18 January 2015.-Instructions for Continued Airworthiness, Northwest Aerospace.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise or effect the performance of the FAA approved product.</td>
<td>03/02/2017</td>
<td>Active</td>
<td>10051947</td>
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<td>EASA STC ref</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10051948</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA 3600/97</td>
<td>BOEING 767-300</td>
<td>Installation of a Broue Aerospace Cabin lighting &quot;Mood Lighting&quot;system. This STC installs new cabin lighting, with LED modules which require less power. These lights for general illumination and provide no emergency lighting. They require manual offshore with a new control panel to allow crew to...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Data Engineering Master Data List, ref: 0144-08509-2368 Revision A dated 11 April 2014 Delta Engineering Instruction for Continued Airworthiness, ref: 0744-08510-2396 Revision A dated 11 April 2014 Bruce Aerospace Weight &amp; Balance report, Ref: 251-3732 Issue B, dated 04 April 2014 FAA AMOC 1405.14-89...</td>
<td>None</td>
<td>20/01/2015</td>
<td>Active</td>
<td>10051948</td>
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<tr>
<td>10051957</td>
<td>AIR ALLIANCE GmbH</td>
<td>REV. 1</td>
<td>35, 35A</td>
<td>Installation of Dual Garmin GTN 750(Xi) NAV/COM and Sandel SN4500 EHSI</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Master Document List, Air Alliance GmbH document No. 2014-001-MDL, Revision 01, dated 28 June 2021 Master Document List, Air Alliance GmbH document No. 2014-001-MDL, Revision 01, dated 26 February 2021 Instructions for Continued Airworthiness, Air Alliance GmbH document No. 2014-001-CA-3</td>
<td>Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/03/2021</td>
<td>Active</td>
<td>10051957</td>
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<tr>
<td>10051958</td>
<td>AIR ALLIANCE GmbH</td>
<td>REV. 2</td>
<td>35A, 35, 35A</td>
<td>SOFTWARE UPGRADE FOR GARMIN GTS 8000 TCAS and GTX 3000 XPD</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Service Bulletin – GTX 3000 and QS 8000 Software Update Rev 2014-005-SB-001 initial issue; Master Document List Ref 2014-005-MDL Initial Issue; latest revision of the above listed document(s) approved/applicable under the EASA system.</td>
<td>Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/05/2023</td>
<td>Active</td>
<td>10051958</td>
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<tr>
<td>10051962</td>
<td>AIR ALLIANCE GmbH</td>
<td>REV. 1</td>
<td>35A, 35A</td>
<td>Installation of dual Rockwell Collins AHS-1000</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List, 2014_004__L335__Form2, revision 02 dated 16 January 2015 Air Alliance, Aircraft Flight Manual, 2014_004__L335_050, revision 02 dated 08 May 2014 Air Alliance, Instructions for Continued Airworthiness, 2014_004__L335_0507 revision 02 dated 09 May 2014 or later revi...</td>
<td>None</td>
<td>04/05/2015</td>
<td>Active</td>
<td>10051962</td>
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<tr>
<td>10051963</td>
<td>AEROTECH &amp; CONCEPT</td>
<td>737-800</td>
<td>737-800</td>
<td>Dual Indium / GPS Antenna Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.</td>
<td>-Master Data List, Aerotec Concept document No. AE14230MDL, Revision 0 dated 21 January 2015 - Instructions for Continued Airworthiness, Aerotec Concept document No. AE14230MEA, Revision 0, dated 18 November 2014 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
<td>21/01/2015</td>
<td>Active</td>
<td>10051963</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>STC TWENTY ONE LIMITED</td>
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<td>Introduction of T2CAS product software</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List: STC Twenty One document No. 521.MOL-1060, issued, 22 January 2015-Fight Manual Supplement, STC Twenty One document no. 521.FMS-015, issue 1, 28 November 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a reduction effect upon the airworthiness of the product.</td>
<td>23/01/2015</td>
<td>Active</td>
<td>10051010</td>
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<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>REV. 1</td>
<td>Installation of ELT antenna and provisions for ELT beacon A06V2.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for the environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List 01437- EASA-A-MDD- rev WiFight and Balance analysis 01437-EASA-A-VSA-01 rev AMPD Supplement 01437- EASA-A-MPD-01 rev Ace later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a reduction effect upon the airworthiness of the product.</td>
<td>06/02/2015</td>
<td>Active</td>
<td>10051060</td>
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<td>ARMSTRONG AEROSPACE Inc.</td>
<td>REV. 1</td>
<td>A321-213/-221/-232 A330-211/-212,-213,-214 A320-211/-221/-232 A330-211/-212,-214 A319-111/-120-133 A319-111/-120-113-115</td>
<td>EASA Validation of FAA STC ST03406CH, originally issued December 18th 2014 Certification of Compatibility with Cellular and Wireless Data Technologies in accordance with Armstrong Aerospace Master Data List Doc. No. DC104-9619-00, Revision, dated December 4th 2014 or later approved revisions.</td>
<td>Armstrong Aerospace Master Data List Doc. No. DC104-9619-00, Revision., dated December 4th 2014 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/0F (or subsequent revisions of this decision) and/or the Technical Implementation Procedures o.</td>
<td>26/01/2015</td>
<td>Active</td>
<td>10051092</td>
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<td>PATS AIRCRAFT L.L.C.</td>
<td></td>
<td>FAA STC ST02853NY</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for the environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List, PATS Aircraft LLC document No. 5976-0001-MDL, Revision P(4)/P(5) dated July 2, 2010 Maintenance Manual Supplement, PATS Aircraft LLC document No. T-0-3317(4)/P(&gt;)/Revision IR, dated May 15, 2010- Airplane Flight Manual Supplement, PATS Aircraft LLC document No. 204, Revision R, date...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a reduction effect upon the airworthiness of the product.</td>
<td>26/01/2015</td>
<td>Active</td>
<td>10051090</td>
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<td>JET AVIATION AG</td>
<td>REV. 1</td>
<td>LEARJET 31/31A/35/35A/36/36A</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Jet Aviation, Aircraft Flight Manual Supplement, TCAS II (ACAS II)Upgrade to Version 7.1, JSSL- M-00342-4FM-01, issue 1Jet Aviation, Instructions for Continued Airworthiness JSSL- M-00342-JCA-01, issue 1, dated 05 January 2015.Jet Aviation, Engineering Order JSSL-M-00342- ENO-01, issue 2, dated 23Janu...</td>
<td>None</td>
<td>02/02/2015</td>
<td>Active</td>
<td>10050011</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10052018</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>8757-200</td>
<td></td>
<td>Installation of a Boeing Aerospace Cabin Lighting Thia STC installs new cabin lighting, with LED modules which rerequire no power. These lights are for general illumination and provide navigation lighting. They require no power and interface with a new control panel to allow crew to change the state of lighting.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, FAA STC 10052018, Rev. A (Certification Plan) or MDL 1005201B, Iss. 1</td>
<td>None</td>
<td>01/12/2015</td>
<td>Active</td>
<td>10052018</td>
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<td>10052021</td>
<td>DART AEROSPACE Ltd.</td>
<td>REV. 1</td>
<td>TCCA STC SH14-45 ISSUE 1</td>
<td>AW119MKII</td>
<td>D-119-796 Crossover Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List, MDL-D119-796, Rev. A. Certification Plan for Installation of above listed documents approved by EASA inaccordance with EASA Decision EASA ED Decision EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)</td>
<td>None</td>
<td>13/02/2015</td>
<td>Active</td>
<td>10052021</td>
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<td>10052022</td>
<td>ARMSTRONG AEROSPACE Inc.</td>
<td>FAA STC ST02276CH</td>
<td>B727-200F</td>
<td>Installation of an ELTA ADT 406 AF</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Conditions: H-01 E124 Transactions: The following paragraphs apply to the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Tec...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce an adverse effect upon the airworthiness of the original product.</td>
<td>27/01/2015</td>
<td>Active</td>
<td>10052022</td>
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<td>10052023</td>
<td>JP INSTRUMENTS, INC</td>
<td>FAA STC SA2586NM</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of a J. P. Instruments temperature monitoring system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Instruments Drawings List Report No. 100, Revision D, dated December 19, 1996 Flight Manual Supplement no. 1 for EG-701 temperatures via FAA STC SA2586NM. Prior to installation of this design change it must be determined that the interrelationship between this design change and any...</td>
<td>Approval of this change in type design to apply to the above model/craft only. The STC requires the installation of the JPI temperature monitoring system via FAA STC SA2586NM. Prior to installation of this design change it must be determined the interrelationship between this design change and...</td>
<td>02/02/2015</td>
<td>Active</td>
<td>10052024</td>
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<tr>
<td>10052045</td>
<td>JP INSTRUMENTS, INC</td>
<td>FAA STC SA00432SE</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Fuel flow transducer</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>J.P. Instruments (JPI) Fuel Flow Installation Manual Report 503, Revision B, dated March 14, 1997. JPI Drawing List, Report No. 560, Revision B, dated 14 March 1997. Approval of this change in type design to apply to the above model/craft only. The STC requires the installation of the JPI fuel flow instrument via STC SA00432SE. Prior to installation of this design change it must be determined that the interrelationship between this design change and any...</td>
<td>Approval of this change in type design to apply to the above model/craft only. The STC requires the installation of the JPI fuel flow instrument via STC SA00432SE. Prior to installation of this design change it must be determined that the interrelationship between this design change and any...</td>
<td>04/02/2015</td>
<td>Active</td>
<td>10052045</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>LITEAIR AVIATION PRODUCTS Inc.</td>
<td>10052046</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Validation of the use-case of the FAA STC No. ST00691SE, dated 20 May 2014 Installation and Maintenance of lightweight window plugs in accordance with Litair Aviation Products Master Drawing List reference LAP1003-00/D0 Revision B, dated March 2014 History. The replacement window plug installation, the a...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Litair Aviation Products Master Drawing List reference LAP1003-00/D0 or subsequent revisions approved by EASA if applicable with EASA STC Decision 2004/04/CF (or subsequent revisions offsideasociation) and/or the Implementation Procedures</td>
<td>Maintenance Instructions: Litair document LAC-003 Rev B dated March 2014 or later approved revision Section 4 details maintenance instructions and these should be incorporated into the aircraft maintenance manual and/or the Programme Limitations. Lightweight window plug installation is not pe...</td>
<td>24/06/2015</td>
<td>Active</td>
<td>10522040</td>
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<tr>
<td>DUNLOP AIRCRAFT TYRES LIMITED</td>
<td>10052047</td>
<td>CL-600-2D4Q CL-600-2D15 CL-600-2C10</td>
<td>Amendment of UKCAA STC reference CAA.21/1102:00100 with partnumbers DR32111T and DR32211T to include a new Main Landing Gear Tyre path number (DR32112T) and a new aircraft variant.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Dunlop Aircraft Tyres Master Document List No. DALT01/7 Issue Initial</td>
<td>None</td>
<td>09/08/2016</td>
<td>Active</td>
<td>10522047</td>
<td></td>
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<tr>
<td>DOA AESGLOBAL LIMITED</td>
<td>10052048</td>
<td>767-300</td>
<td>Introduction of 258 Passenger Configuration and Associated Cabin Configuration and Associated Cabin Change (Rev. 1) Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>Aerospace Engineering Solution Master Data List AES-787-329MDL at Issue 3 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>28/04/2021</td>
<td>Active</td>
<td>10522049</td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>10052049</td>
<td>A321-211/231 A320-214/230/233 A319/112/132</td>
<td>Wireless In-Flight Entertainment (IFE) System Installation on LATAM Aircraft</td>
<td>-Certification Summary, Northwest Aerospace Technologies, Inc. document N056-D09, Revision A, dated 11 September 2015. -Instructions for Continued Airworthiness, Northwest Aerospace Technologies, Inc. document No. N494- D10, Revision C, dated 11 May 2015 or later revisions of the above listed document(s)</td>
<td>Aerospace Engineering Solution Master Data List AES-787-329MDL at Issue 3 or later approved revision</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/10/2015</td>
<td>Active</td>
<td>10522049</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>10052055</td>
<td>BAE 146 SERIES 300 BAE 146 SERIES 200 BAE 146 SERIES 100 AVRO 146 SERIES RJS AVRO 146 SERIES RJ70 AVRO 146 SERIES RJ15 AVRO 146 SERIES RJ100</td>
<td>Introduction of TCAS Change 7.1</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements.</td>
<td>&quot;EASA Approved&quot;: STC Twenty One Flight Manual Supplement S21.FMS-0116Issue 1 &quot;EASA Accepted&quot;: STC Twenty One Master Document S21.MD.1221 Issue 2</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/01/2015</td>
<td>Active</td>
<td>10522055</td>
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<tr>
<td>STC TWENTY ONE LIMITED</td>
<td>10052056</td>
<td>3. 58TC. 58TCA 2. 58. 58A 1. 36. 36A. 36TC. 36BTC</td>
<td>Installation of a stretcher on to the seat rails.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Aviation Fabricators Drawing List ref to AF-162 dated May 25, 1993 and Installation Instructions ref AF-116 dated 12/28/93 Maintain in accordance with Aviation Fabricators Instructions for Continued Airworthiness (ICA) Document Ref N0 AF-609 Rev IR dated October 6, 2014.</td>
<td>Whilst this installation provides some provisions for carriage of medical equipment. This STC grants no approval to installation of primary carry-on medical equipment. Any equipment fitted/certified will require as acceptable EASA approval. Prior to installation of this design change it must be determined that the...</td>
<td>29/01/2015</td>
<td>Active</td>
<td>10522075</td>
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<tr>
<td>AVIATION FABRICATORS, Inc.</td>
<td>10052075</td>
<td>FAA STC SA005891</td>
<td>Installation of a stretcher on to the seat rails.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Aviation Fabricators Drawing List ref to AF-162 dated May 25, 1993 and Installation Instructions ref AF-116 dated 12/28/93 Maintain in accordance with Aviation Fabricators Instructions for Continued Airworthiness (ICA) Document Ref N0 AF-609 Rev IR dated October 6, 2014.</td>
<td>Whilst this installation provides some provisions for carriage of medical equipment. This STC...</td>
<td>29/01/2015</td>
<td>Active</td>
<td>10522075</td>
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<td>STC Holder</td>
<td>FAA STC</td>
<td>Rev. level</td>
<td>Foreign/NAAL</td>
<td>STC ref</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>ANJOU AERONAUTIQUE S.R.L.</td>
<td>ST0214</td>
<td>10052077</td>
<td></td>
<td>A320-214</td>
<td>Cabin reconfiguration</td>
<td>Master Data List, Anpuahro document No. 02-401888-MDL-001 Revision 1,dated28&lt;sup&gt;th&lt;/sup&gt; January 2015&lt;sup&gt;th&lt;/sup&gt;Airplane Flight Manual Supplement, Anpuahro document No. 02-401888-AFMS-001, Revision 0,dated 18&lt;sup&gt;th&lt;/sup&gt; November 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>The design change is limited to MSN 3025. Extended overwater operations are prohibited.</td>
<td>29/01/2015</td>
<td>Active</td>
<td>1052077</td>
<td></td>
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<tr>
<td>PMV ENGINEERING</td>
<td>ST0200</td>
<td>10052078</td>
<td></td>
<td>767-200, 767-300</td>
<td>B767 Cabin interior change and In-Flight Entertainment (IFE) systeminstallation;The modification consists in the reconfiguration of the cabin from a two class 221 PAX (30 B/C + 164 Y/C) to a three class 221 PAX (30 B/C + 30 B/C + 164 Y/C), together with the installation of an IMS2odiac Infightimrnovatio...</td>
<td>Master Document List, PMV Engineering document SCA-0320-MDL-R20 dated29&lt;sup&gt;th&lt;/sup&gt; January 2015&lt;sup&gt;th&lt;/sup&gt;Certification Investigation Report, PMV Engineering document SCA-0320-CIR-R05 enclosing Declaration of Compliance dated 29 January2015&lt;sup&gt;th&lt;/sup&gt;Certification Investigation Report, PMV Engineering document CAB-01030-CIR-R05.</td>
<td>EASA, The following design changes are considered as part of the prerequisite reconfiguration of the aircraft and must be embodied on aircraft prior, or concurrently, to the application of the design change covered by this approval. STC FAA ST03665AT: Lavatories installation change- STC FAA ST0214NY: Retractable CabinNolat...</td>
<td>29/01/2015</td>
<td>Active</td>
<td>1052078</td>
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<tr>
<td>GENEVA AVIATION Inc.</td>
<td>ST0201E</td>
<td>10052094</td>
<td></td>
<td>EC 135 T1, EC 135 T2, EC 135 P2+ EC 135 T2+ EC 135 P1, EC 135 P2</td>
<td>Installation of Geneva Aviation 28-14 VDC Power Converter per Geneva Aviation Master Drawing List (MDL) GA203</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise previously installed design change or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Certificate ST024811A</td>
<td>30/01/2015</td>
<td>Active</td>
<td>1052094</td>
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<tr>
<td>TASS Inc.</td>
<td>ST0419M</td>
<td>10052096</td>
<td></td>
<td>737-400</td>
<td>Installation of Dual FMS and Dual GPSs</td>
<td>Tracono Master Drawing List (MDL) 97-001, Sheets 1, 2 and 3;Revision A dated1&lt;sup&gt;st&lt;/sup&gt; February 1997.</td>
<td>None</td>
<td>30/01/2015</td>
<td>Active</td>
<td>1052096</td>
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<td>ASI INNOVATION</td>
<td>ST0100</td>
<td>10052105</td>
<td></td>
<td>ERJ 190-100 LR</td>
<td>Embraer 190 Cabin Modification to 110 Passengers</td>
<td>ASI STC Data List ref. ASI- SSL-0220-02 dated 08&lt;sup&gt;th&lt;/sup&gt; December 19&lt;sup&gt;th&lt;/sup&gt;earlier revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any otherwise previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/02/2015</td>
<td>Active</td>
<td>1052105</td>
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<td>STC Holder</td>
<td>FAA STC</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AEROLITE AG</td>
<td>ST02214Y-D</td>
<td>Interior Reconfiguration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental/protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Greater Aviation Master Drawing List (MDL) GA203</td>
<td>None</td>
<td>02/02/2015</td>
<td>Active</td>
<td>10052125</td>
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<tr>
<td>SAFRAN CABIN BELINGHAM, INC.</td>
<td>BO 105 S</td>
<td>Installation of Geneva Aviation 28-14 VDC Power Converter per Geneva Aviation Master Drawing List (MDL) GA203</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental/protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Greater Aviation Master Drawing List (MDL) GA203</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2015</td>
<td>Active</td>
<td>10052138</td>
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<tr>
<td>GENEVA AVIATION Inc.</td>
<td>BO 105 A</td>
<td>Installation of Geneva Aviation 28-14 VDC Power Converter per Geneva Aviation Master Drawing List (MDL) GA203</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental/protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Greater Aviation Master Drawing List (MDL) GA203</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2015</td>
<td>Active</td>
<td>10052139</td>
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<tr>
<td>GENEVA AVIATION Inc.</td>
<td>EC 130 B4, EC 130 T2</td>
<td>Installation of Geneva Aviation 28-14 VDC Power Converter per Greater Aviation Master Drawing List (MDL) GA203</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental/protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Greater Aviation Master Drawing List (MDL) GA203</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2015</td>
<td>Active</td>
<td>10052140</td>
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<tr>
<td>GENEVA AVIATION Inc.</td>
<td>AS 355 N, AS 355 NP</td>
<td>Installation of Geneva Aviation 28-14 VDC Power Converter per Greater Aviation Master Drawing List (MDL) GA203</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental/protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Greater Aviation Master Drawing List (MDL) GA203</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2015</td>
<td>Active</td>
<td>10052141</td>
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<td>STC Holder</td>
<td>STC Reference</td>
<td>Rev. or Ed.</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>A380-861</td>
<td>REV. 2</td>
<td>A380-861</td>
<td>650</td>
<td>Activation of On-Board Mobile Telephone</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Supplemental Type Definition Document - ST0213/4/14/05M - Compliance Record List - CRL44-14/02M - AFM Supplement - A-30-04/1480-AFM - Instructions for Continued Airworthiness - A-00-04/1480/ICA-MMEL-L - A-00-04/1480/MMELor later revisions of the above listed documents approved by EASACompliance Document APV...</td>
<td>None</td>
<td>08/10/2015</td>
<td>Active</td>
<td>10052187</td>
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<tr>
<td>AERODATA AG</td>
<td>A380-861</td>
<td>REV. 1</td>
<td>AD-AFIS-0116 on King Air B200</td>
<td>B200</td>
<td>Installation AD-AFIS-0116 on King Air B200</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Document List, AERODATA AG Innovation document No. AS-B3-055/2015 Revision 0, dated 14 February 2015, or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of the design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2015</td>
<td>Active</td>
<td>10052155</td>
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<tr>
<td>ASI INNOVATION</td>
<td>A380-861</td>
<td>REV. 1</td>
<td>ASI-SDL-0240</td>
<td>550</td>
<td>Certification of P-RNAV function of aircraft equipped with Garmin GNS530AW</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Document List, ASI Innovation document No. ASI-SDL-055/04 Revision 0, dated 14 October 2014, or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of the design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/02/2015</td>
<td>Active</td>
<td>10052170</td>
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<tr>
<td>AIRBORNE MAINTENANCE ANDENGINEERING SERVICES, Inc.</td>
<td>FAA STC ST0103/08</td>
<td>REV. 1</td>
<td>767-200, 767-300, 767-100</td>
<td>400W High Frequency (HF) Voice Radio System ins...</td>
<td>Installation of dual Honeywell X516D 400W High Frequency (HF) Voice Radio Systems - configuration 1: installation of provisions on aircraft without HF Provision or aircraft provisioned for fury kit configuration 2: installation of provisions on aircraft with a single HF Voice Radio System ins...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (a)(4) EWIS: The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged...</td>
<td>- Master Data List, RPT-04-07311 rev LInstructions for Continued Airworthiness RPT 04-0731/5 rev CAPM Supplement 38033/AFMRev1R0or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical...</td>
<td>Prior to installation of the design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/02/2015</td>
<td>Active</td>
<td>10052171</td>
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<tr>
<td>LUFTHANSA TECHNIC AG</td>
<td>A380-861</td>
<td>REV. 2</td>
<td>A380-861</td>
<td>Activation of On-Board Mobile Telephone</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>None</td>
<td>- Supplemental Type Definition Document - ST0213/4/14/05M - Compliance Record List - CRL44-14/02M - AFM Supplement - A-30-04/1480-AFM - Instructions for Continued Airworthiness - A-00-04/1480/ICA-MMEL-L - A-00-04/1480/MMEL or later revisions of the above listed documents approved by EASACompliance Document APV...</td>
<td>None</td>
<td>08/10/2015</td>
<td>Active</td>
<td>10052167</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>STC Holder AEROSERVIS, (MEDAVIA)</td>
<td>REV. 1</td>
<td>10052196</td>
<td>1900C AND 1900D</td>
<td>Introduction of TCAS II Version 7.1 on Beechcraft 1900 C and D Series.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate approval.</td>
<td>TCAS 20000 Upgrade Installation in accordance with Medavia Service Bulletin net/MB89131/197/B_T Bearing 1900 C and D Series.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a tradeoff adverse effect upon the airworthiness of the product.</td>
<td>06/10/2015</td>
<td>Active</td>
<td>10052196</td>
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<td>TESTORI AERO SUPPLY S.R.L.</td>
<td></td>
<td>10052197</td>
<td>737-900ER</td>
<td>Flight Attendant Seat Cushion Replacement for Boeing 737 NGseries</td>
<td>None</td>
<td>Component Maintenance Manual and IPC TAS-STC-B737NG-FA-CM/MA/187 prior revisions of the above listed documents approved by EASA Master Document List TAS-STC-B737NG-FA-CM/MA/187 Rev. 00/01 Master Drawing List TAS-STC-B737NG-FA-CM/MA/01 Rev. 00</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this change and any other previously installed design change will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/02/2015</td>
<td>Active</td>
<td>10052197</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>10052215</td>
<td>AS 332 L2</td>
<td>Installation of Long Wire Antenna and Activation of KNA-559HF Radioloc System</td>
<td>The Certification Basis for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate approval.</td>
<td>Bouremouth Aviation (Consultants) Rotorcraft Flight Manual Supplement BAC-937MF01 Revision 1 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a tradeoff adverse effect upon the airworthiness of the product.</td>
<td>10/02/2015</td>
<td>Active</td>
<td>10052215</td>
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<tr>
<td>KITCHENER AERO AVIONICS, Ltd.</td>
<td>TCCA STC SHU-2-61</td>
<td>10052221</td>
<td>AS 332 L1</td>
<td>Installation of Garmin G500H Flight Display System and simultaneous replacing of G500 and G500P 32 Air Data Display, initial revision</td>
<td>The Certification Basis for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate approval.</td>
<td>Bouremouth Aviation (Consultants) Rotorcraft Flight Manual Supplement BAC-937MF01 Revision 1 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a tradeoff adverse effect upon the airworthiness of the product.</td>
<td>10/02/2015</td>
<td>Active</td>
<td>10052221</td>
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<tr>
<td>AEROSERVIS, S.R.O.</td>
<td>L 410 UVP-E, -E9, -E20</td>
<td>10052228</td>
<td>A330-300</td>
<td>Introduction of Enhanced Surveillance to existing Mode S Transponder MTS67A as per document AE153 001P.</td>
<td>The Certification Basis for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this...</td>
<td>STC AE153 Master Document List, Doc. No. AE153000X, rev D(706) dated 10.12.2014. 410 UVP-E Flight Manual Supplement Doc. No. STC AE153-000X dated 10.12.2014. 410 UVP-E Flight Manual Supplement Doc. No. STC AE153000X dated 10.12.2014.</td>
<td>The installer must determine whether this design change is compatible with previously approved modifications if the holder approves to permit another person to use this certificated product, the holder shall give the other person entitlement of that permission.</td>
<td>20/02/2015</td>
<td>Active</td>
<td>10052228</td>
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<tr>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 2</td>
<td>10052233</td>
<td>A330-300</td>
<td>SAS A330-343 Cabin configuration/Replacement of Business, Premium Economy and Economy Class seats, galley, stairs, IFE and ITPSS. Relocation of main cabin lavatories, removal of cabin crew rest module Rev. 2. Transfer of STC from SR Technics Switzerland to geWING Engineering Partners GmbH</td>
<td>The Certification Basis for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this...</td>
<td>Aircraft Flight Manual Supplement S.OA.D 00300-ARM, rev 00 Master Document List S.OA.D. 00300-ARM, rev 00. Instructions for the Continuous Airworthiness S.OA.D 00300-ICA, rev 01. For latemorons of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a tradeoff adverse effect upon the airworthiness of the product.</td>
<td>20/04/2021</td>
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<td>AVIONTEC AG</td>
<td>10052261</td>
<td></td>
<td></td>
<td>M20C</td>
<td>Equipment Upgrade incl. AFMS Revision for IFR Operation. Out: Britain autopilot system, Bendix/King DME KN-84, Bendix/King NAV Indicator KI 203, Garmin GNS-430 system In: Bendix/King KN-62A, Garmin GNS-430W system</td>
<td>Certification Basis for the original product as amendedly or alternatively airworthiness requirements: the following paragraph(s) at a later amendment FAR23 Amendment 55; 23-301.561, 1301, 1309, 1321, 1327, 1329, 1351, 1357, 1359, 1365, 1381, 1431, 1529, 1541, 1581 The requirement...</td>
<td>AFM Supplement GNS 430W No. 4445-011 Rev. 1 of 06.01.2015 AFM Supplement PowerFlarm No. 4445-012 Original Issue of 06.01.2015 for later revisions of the above listed documents approved by EASA Master Document List No.4445-017 Rev. 2 of 02.02.2015, Instructions for Continued Airworthiness No. 4445-015.</td>
<td>Limited to S/N 3269 HS-DWOPrior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/02/2015</td>
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<td>JAMCO AMERICA, INC.</td>
<td>10052263</td>
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<td></td>
<td>737-900ER</td>
<td>Installation of Flight Deck Security Door</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>For applicability of Airplane Flight Manual Supplements refer to FAA STC No. ST01435E or later revisions of the above listed documents approved by EASA.Jamco America, Inc Master Drawing List (MDL) IA-0650COMMDL Revision A2 dated 01 December 2014. For applicability of ICA per aircraft serial number...</td>
<td>Installation of this STC replaces the original cockpit door with a new security flight deck door and is approved as an Alternate Method of Compliance (AMOC) to Paragraph A and B of Airworthiness Directive (AD) 89-04-03. Approval of this AMOC is documented in FAA Letter 1505-14-13, dated 15 October 2014.</td>
<td>13/02/2015</td>
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<td>JET AVIATION FLUGZEUGWART UNG GmbH</td>
<td>10052268</td>
<td></td>
<td></td>
<td>650</td>
<td>Certification of P-RNAV capability of aircraft equipped with Flight Management System Universal UNS-1D</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>- Master Document List, Cessna Düsseldorf document No. 2014-9-15, Revision (R), dated 12 May 2014. - Airplane Flight Manual Supplement, Cessna Düsseldorf document No. 2014-5-8, Revision (R), dated 12 May 2014 for later revisions of the above listed documents approved by EASA. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/02/2015</td>
<td>Active</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>10052269</td>
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<td></td>
<td>BEECH B300/B300C</td>
<td>Installation of UNIFIS 3000 Flight Inspection System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Scandinavian Avionics Design Engineering Order ref 2078097/E0-01(&gt;,&lt;) Issue 1(&gt;,&lt;) dated 28.01.2015 and Master Drawing List ref 2078097-20(&gt;,&lt;) Issue 1(&gt;,&lt;) dated 17.03.2014. Operate in accordance with Scandinavian Avionics Design Flight Manual Supplement ref 2078097...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/02/2015</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>10052272</td>
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<td>737-900ER</td>
<td>Introduction of TCAS Change 7.1</td>
<td>The Certification Basis for the original product as amendedly or alternatively airworthiness requirements: the following paragraph(s) at a later amendment CS25 Amendment 190C 25.1301; CS 190C 25.1309; CS 25.130C3CS 25.1431; CS 25.1581; CS 25.1585 The requirements for environmental protection...</td>
<td>STC Twenty One Flight Manual Supplement S21.FSM-0117 Issue 1 for later revisions of the above listed documents approved by EASA. All of the documents detailed in STC Twenty One Master Document List S21.MDL-1225 Issue 2. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/02/2015</td>
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<td>DUESSLEDORF CITATION SERVICE CENTER GmbH</td>
<td>10052273</td>
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<td>P 180 AVANTI</td>
<td>P-RAU Capability within Universal UNS-1/J/H Flight Management System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airdrome Flight Manual Supplement UNS-1L, 2014-14, Rev.</td>
<td>None</td>
<td>13/02/2015</td>
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<td>CAA UK RESPONSIBIL ITY</td>
<td>10052279</td>
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<td>CL600-2B16</td>
<td>Introduction of a Corporate Interior</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>STC Twenty One Flight Manual Supplement No. S21.FMS-002 Issue 1 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>17/02/2015</td>
<td>Active</td>
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<td>BERINGER AERO</td>
<td>10052286</td>
<td>REV. 5</td>
<td>EA 300/SC EA 300/LT, EA 300/S EA 300L, EA 300/LC EA 300, EA 300/200</td>
<td>Replacement of wheel and brake systems: Rev. 1 - Update with longer axle bolts/rev. 2 - Update with different hydraulic fittings/rev. 3 - Test Report Improvement/rev. 4 - Update with new wheel part number/rev. 5 - Update with new servicing instructions - Additional of SensAir device as an opt.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification file: DM-STC-012 rev. 06 dated 11 Mar. 2022. Installation and Instructions for Continued Airworthiness. ICA-STC-012 rev. 04 dated 19 Oct. 2021 (or later revisions of the above listed documents approved by EASA).</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 21.A.109 or 21.A.451 (b) (as applicable). Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce unrepairable effect upon the airworthiness of the product.</td>
<td>18/03/2022</td>
<td>Active</td>
<td>10522299</td>
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<tr>
<td>DART AEROSPACE Ltd.</td>
<td>10052312</td>
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<td>SH14-44</td>
<td>D066-1004 Cargo Hook Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement FMS-D066-1004 Revision B, dated 17 Feb. 2015 (or later revisions of the above listed documents approved by EASA).</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce unrepairable effect upon the airworthiness of the product.</td>
<td>18/02/2015</td>
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<td>10523112</td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>10052319</td>
<td>REV. 2</td>
<td>A330-2011-2021-203</td>
<td>Cabin Interior Reconfiguration QANTAS Airways Aircraft</td>
<td>The Certification Basis for the original product as amended is applicable to the following additional or alternative airworthiness requirements: Special Condition (s): CRI-D-3327-202 Incorporation of ILD in Dynamic/Static. The following paragraph(s) at a later amendment: CS 25.776(1)(a) as amended 21.A.109. The requirements for e...</td>
<td>Certification Summary, Northwest Aerospace Technologies, Inc. document N0453-D01. Revisions, dated 30 November 2018. Instructions for Continued Airworthiness, Northwest Aerospace Technologies, Inc. document N0453-D07, Revision A, dated 9 December 2014 (or later revisions of the above listed documents).</td>
<td>Monitors and components of the IFE System, including In-Seat PowerSupply System are treated as pure items of mass under the approval. Electrical connection to aircraft requires further design approval including full equipment qualification. Prior to installation of this change repair it must be...</td>
<td>29/04/2019</td>
<td>Active</td>
<td>10523119</td>
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<tr>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>10052320</td>
<td>REV. 1</td>
<td>FAA ST03357CH</td>
<td>737-600-700-800-900/9-00ER</td>
<td>Electronic Flight Bag installation: Configuration A - “Class 3” installed EFB system-Configuration B - “Class 2” EFB host platform (“proverson”</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master data list D0641567 Rev BFS (V3) report D0641547 rev D0641532 rev BConfiguration A, AFM Supplement D0641906 Rev A, Instructions for continued airworthiness are D0642332 rev CConfiguration B, AFM Supplement D06419438 Rev A, Instructions for continued airworthiness are D06419460 rev Cor later revisions of the ab...</td>
<td>This airworthiness approval does not replace the operational assessment for the use of the EFB system. Prior to installation of this change repair it must be...</td>
<td>24/02/2015</td>
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<td>MARSHALL OF CAMBRIDGE AEROSPACE LIMITED</td>
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<td>A320-231/232-233 A320-211/212-214</td>
<td>A320 LiveTV Ka Band Wi-Fi Connectivity Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>10052356 MA Supplement No. AFMS/ A320(2022)/14/1068 Rev. 1 to Airbus A320-200 Series Aircraft Flight Manual: Aircraft System Installation of an AVIO Business Class Interior</td>
<td>Must be installed in conjunction with EASA STC 10050344 and in accordance with the limitations of that STC Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/02/2015</td>
<td>Active</td>
<td>10052356</td>
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<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 1</td>
<td></td>
<td>MF 50EX</td>
<td>Interior Modification and In-Flight Entertainment System upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>- Master Document List, 328 Design GmbH document No. MDL0432001, Revision A, dated 2 February 2015; or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/09/2018</td>
<td>Active</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 3</td>
<td></td>
<td>ERJ 170-200 STD ERJ 170-200 LR ERJ 170-100 STD ERJ 170-100 LR</td>
<td>Installation of Photoluminescent Emergency Floor Path Marking System. This approval constitutes the initial Operational Suitability Data MMP/EA (Rev 1).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: All applicable paragraphs according to CS 25 at Amendment 15 (eject tismooy). The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>- EASA STC no. 737-800</td>
<td>A320-200 and A320-300 Series aircraft Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/06/2016</td>
<td>Active</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 3</td>
<td></td>
<td>ERJ 190-200 STD ERJ 190-200 LR ERJ 190-100 STD ERJ 190-100 SR ERJ 190-100 LR ERJ 190-100 IBD ERJ 190-100 ECJ</td>
<td>Installation of Photoluminescent Emergency Floor Path Marking System. This approval constitutes the initial Operational Suitability Data MMP/EA (Rev 1).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: All applicable paragraphs according to CS 25 at Amendment 15 (eject tismooy). The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>- EASA STC no. 737-800</td>
<td>A320-200 and A320-300 Series aircraft Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/03/2016</td>
<td>Active</td>
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<td>ICELANDAIR DESIGN ORGANISATION</td>
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<td>FAA STC ST03340CH</td>
<td>Installation of an AVIO Business Class Interior (ST03340CH)- Installed on an STC for a new Interior and IFE System as defined in Certification Plan GR0050- CPL Rev. 1 dated 02-05-2013</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/419/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EULISA Bilateral Agreement.- Master Data List, Doc. No. GR0051-MDL, Rev. E, dated 05-12-2013</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/02/2015</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td><strong>SABENA TECHNICS</strong></td>
<td><strong>10052395</strong></td>
<td><strong>REV. 3</strong></td>
<td><strong>EC 225 LP</strong></td>
<td><strong>ADS-B OUT in accordance with Master Data List 2043-01-A2103</strong></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 29.Amdt 4 (except § 29.341 at JAR 25) Change 14) The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Aerospatiale Helicopter Flight Manual Supplement (RFMS) No 2043-01-A2103, for later revisions of the above listed documents approved by EASA: Master Data List, Doc. No. 5312-049, Rev. B, dated 19.05.2014: ICA, Doc. No. 5312-085, Rev. A, dated 19.05.2014, Cabi...</td>
<td>- Certification Plan EASA: Master Data List No. 5312-049, Rev. B, dated 19.05.2014</td>
<td>- Certification Plan EASA: Master Data List No. 5312-049, Rev. B, dated 19.05.2014</td>
<td>24/02/2020</td>
<td>Active</td>
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<td><strong>JET AVIATION AG</strong></td>
<td><strong>10052409</strong></td>
<td><strong>REV. 3</strong></td>
<td><strong>737-800</strong></td>
<td><strong>VIP Interior - two kinds of Operation. The change is intended to establish the conditions under which the cabin be changed between &quot;private&quot; and &quot;commercial&quot; configuration based on Conversion Manual, Doc. No. JBSL-M-00129DSS-01, Issue 1, dated 16.02.2016</strong></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Deviation(s) CRI D-01 Interior DoorsCRU D-06 Flight Handrail The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Airbus Helicopter Flight Manual Supplement FMS-135-049898-501 Revision C - Master Document List MDL-14-039</td>
<td>- Certification Plan EASA: Master Data List No. 5312-049, Rev. B, dated 19.05.2014: ICA, Doc. No. 5312-085, Rev. A, dated 19.05.2014, Cabi...</td>
<td>- Certification Plan EASA: Master Data List No. 5312-049, Rev. B, dated 19.05.2014</td>
<td>- Certification Plan EASA: Master Data List No. 5312-049, Rev. B, dated 19.05.2014</td>
<td>25/02/2015</td>
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<td>HELI-ONE (NORWAY) AS</td>
<td>A332L, A332L1</td>
<td>CVFROR Installation in accordance with Modification AHSE-64750.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-29 Amendment 3, 29.1.29: 45-261-301:302:305 / 307:321:337:601:603:605:60 7:606:610:613:625:771:773:1 301...</td>
<td>AT/No.7.237 Revision, Dated 18 February 2015, Certification Programme: AHSE-64755.</td>
<td>- Master Drawing List, EAD Aerospace document No. 01451-EAS-C-ICA-02, Revision A, dated 8 November 2017.</td>
<td>Instructions for Continued Airworthiness, EAD Aerospace document No. 01451-EAS-C-ICA-02, Revision A, dated 8 November 2017. - Aircraft Flight Manual Supplement, EAD Aerospace document No.01451-EAS-C-0-..</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/02/2015</td>
<td>Active</td>
<td>10052415</td>
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<tr>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>A340-311,-312,-313 A340-211,-212,-213</td>
<td>TPED Imunity Testing.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- EASA Approval VH Aerospace Inc Rotorcraft Flight Manual Supplement, VS92-0726-RS-FLIR Rev 1, Spectrolab Nightsun II XPiSarlight System/VH Aerospace Inc Rotorcraft Flight Manual Supplement, VS92-0726-FLIR-FMS, Rev 4, FLIR U8500R Airborne Imaging System &amp; SAR/Operator Console, or later revisions of the above listed documents approved by EASA.</td>
<td>- EASA Approval VH Aerospace Inc Rotorcraft Flight Manual Supplement, VS92-0726-RS-FLIR Rev 1, Spectrolab Nightsun II XPiSarlight System/VH Aerospace Inc Rotorcraft Flight Manual Supplement, VS92-0726-FLIR-FMS, Rev 4, FLIR U8500R Airborne Imaging System &amp; SAR/Operator Console, or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/01/2018</td>
<td>Active</td>
<td>10052424</td>
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<tr>
<td>VIH AEROSPACE Inc.</td>
<td>TCCA SH09-39 ISSUE 2 S-92A</td>
<td>SAR Equipment Installation - Nightsun and FLIR</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- EASA Approval VH Aerospace Inc Rotorcraft Flight Manual Supplement, VS92-0726-RS-FLIR Rev 1, Spectrolab Nightsun II XPiSarlight System/VH Aerospace Inc Rotorcraft Flight Manual Supplement, VS92-0726-FLIR-FMS, Rev 4, FLIR U8500R Airborne Imaging System &amp; SAR/Operator Console, or later revisions of the above listed documents approved by EASA.</td>
<td>- EASA Approval VH Aerospace Inc Rotorcraft Flight Manual Supplement, VS92-0726-RS-FLIR Rev 1, Spectrolab Nightsun II XPiSarlight System/VH Aerospace Inc Rotorcraft Flight Manual Supplement, VS92-0726-FLIR-FMS, Rev 4, FLIR U8500R Airborne Imaging System &amp; SAR/Operator Console, or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/03/2015</td>
<td>Active</td>
<td>10052427</td>
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<tr>
<td>ASI INNOVATION</td>
<td>ATR 42-320</td>
<td>SATCOM System installation on ATR 42/32This modification installs a SATCOM system composed of the following elements: SATCOM Transceiver, Mission Management Unit (MMU), ConfigurationOM module, Diagnostic port, InflightGPS antenna, indicators lights, and the associated wiring harness.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25 Amnd.4: see Compliance Check-List document, ref: ASI-CLL-0078-03, for the applicable requirements.</td>
<td>- Flight Manual Supplement, ref: ASI-SDL-0078-02, dated 25 February 2015. Aircraft Flight Manual Supplement, ref: ASI-APM-0078-01, dated 26 January 2015.</td>
<td>- Flight Manual Supplement, ref: ASI-SDL-0078-02, dated 25 February 2015, Aircraft Flight Manual Supplement, ref: ASI-APM-0078-01, dated 26 January 2015.</td>
<td>- Flight Manual Supplement, ref: ASI-SDL-0078-02, dated 25 February 2015, Aircraft Flight Manual Supplement, ref: ASI-APM-0078-01, dated 26 January 2015.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/02/2015</td>
<td>Active</td>
<td>10052430</td>
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<tr>
<td>LUTHANSA TECHNIK AG</td>
<td>A380-841</td>
<td>New Premium Economy Class for A380-800The main cabin of the main floor on the aircraft, the change consists the removal of 10 per cent of the Economy Class Seats between door 2and door 3 and the installation of new developed Premium Economy Seat installed (with Panasonic IFE). The new seating will...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Supplemental Type Definition Document STDO25-123BM issue 1 rev 0-0.</td>
<td>- Compliance Record List CRL25-123BM issue 1 rev 1.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/02/2015</td>
<td>Active</td>
<td>10052436</td>
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<tr>
<td>DOA21 LIMITED</td>
<td>FAA STC SA02360AK</td>
<td>1. A-1C-200 2. A-1-1A, A-1B, 1-1C-180 1. 180U, 180K 1. 180E, 180G, 180K, 180H 1. 180, 180A, 180B, 180C, 180D</td>
<td>Fabrication and Installation of Airglas, Inc., Model GLH3000 hydraulically retractable wheel skis. The Certification Baseline for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>STC TWENTY ONE FLIGHT MANUAL SUPPLEMENT No. 21.1305-344</td>
<td>1. EASA STC No.10035344 is a prerequisite for this STC.2. The new airworthiness limitations for the type of operation &quot;Private Category&quot; only, the maximum passenger capacity of 27, and for the newrearg video monitor to be in the retracted (lowered) position for taxitake-off and landing are to be...</td>
<td>24/10/2023</td>
<td>Active</td>
<td>10052458</td>
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<td></td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>A380-841</td>
<td>TCAS Change 7.1 on Boeing DC9/MD80. The Certification Baseline for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TCAS Change 7.1 on Boeing DC9/MD80 TCAS II Flight Computer installed in place of the existing Honeywell TPA-81A computer in the same location. The original tray no. TPA-81A is fully compliant with ETSO-119c including both RTCA/ DO-188B &amp; RTCA II change 7.1.</td>
<td>H4ACPL000, Revision 4, or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU-Canada Bilateral Agreement.</td>
<td>02/02/2015</td>
<td>Active</td>
<td>10052449</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>DISCOVERY AIR TECHNICAL SERVICES</td>
<td>10052479</td>
<td>RH Forward Galley Drawer Insert</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product ar unchanged and remain applicable to this certificate/ approval.</td>
<td></td>
<td></td>
<td>04/03/2015</td>
<td>Active</td>
<td>10552494</td>
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<td>FAA STC</td>
<td>10052484</td>
<td>Certification of Compatibility with Cellular and Wireless Data Technologies Refer also to Envoy Aerospace Certification Plan, Document E13047-2330-C001 rev -</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative environmental requirements: 25.1309(a) at CS amendment 225.1529 at CS amendment 27 The requirements for environmental protection and the associated certified noise and/or emissions levels of the origin...</td>
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<td>04/03/2015</td>
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<td>10552494</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10052493</td>
<td>G280 FANS-1A STC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product ar unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification Plan G280-GER-0013, Rev. - Date: 5th of June 2012(AFMS G280-GER-025A, Rev 3rd of October 2015 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions off this decision) and/or the Technical Implementation Plan...</td>
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<td>05/03/2015</td>
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<td>10552493</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10052494</td>
<td>Installation of an International Communications Group (ICG) SATCOM System and Antennas</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product ar unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification Plan GA44103039C001, Rev A, Dated 11th of November 2014.AFMS GA44103039M001, Rev A, Dated 24th of March 2014 later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions off this decision) and/or the Technical Implementation Plan...</td>
<td>WLAN capability of the portable terminals installed is not covered by this approval.</td>
<td>05/03/2015</td>
<td>Active</td>
<td>10552494</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10052495</td>
<td>Installation of a second Integrated Flight Information System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product ar unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification Plan GA4Y112003C001, Rev. G, Dated 30th of April 2014.AFMS GA44112003M001, Dated 12th of October 2013 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions off this decision) and/or the Technical Implementation Plan...</td>
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<td>02/04/2015</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<tr>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA STC ST04271AT-D</td>
<td>G280</td>
<td>Installation of a Jumpseat in G280.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate / approval.</td>
<td>Certification Part GA42132DC001, Rev F, Dated 4th of January 2014.AFM8 GA42132MD001: +/+/ Rev A, Dated 3rd of November 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/94/CF (or subsequent revisions of this decision) and/or the Technical.</td>
<td>None</td>
<td>05/03/2015</td>
<td>Active</td>
<td>10052498</td>
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<tr>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>REV. 1</td>
<td>FAA ST04286AT-D</td>
<td>GULFSTREAM G280</td>
<td>Installation of an Executive Passenger Cabin Interior Rev. 1: New part numbers for life rafts.</td>
<td>The Type and OSF Certification Bases for the original product are amended by the following additional or alternative airworthiness requirements: Special Condition(s): FAA special condition 25-694-SC. The requirements for environmental protection and the associated certified noise and/or emissions levels.</td>
<td>Index List GA42204B100, Rev. STAFMS GA42204M000, Rev. RInstructions for Continued Airworthiness GA42204A000, Rev. No. later revisions of the above listed document(s) approved / accepted on behalf of EASA in accordance with the Technical Implementation Procedures of the USA-Bilateral Agreement.</td>
<td>06/02/2023</td>
<td>Active</td>
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<td>CL-600-2B16 (601-3A)</td>
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<td>A321-213/-211/-214/-215/-216</td>
<td>A320-211/-212/-214/-215/-216</td>
<td>A319-131/-132/-133</td>
<td>A319-111/-112/-113/-114/-115</td>
<td>Installation of Water Vapour Sensor System II</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate / approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate / approval.</td>
<td>Supplemental Type Definition Document STD000-1419M Issue 2Rev. 0, dated 05/03/2015, Compliance Record List CRL00-1419M Issue 2 Rev. 0, dated 05/03/2015, Instruction for Continued Airworthiness A-00-1419-JCA Issue 2 Rev. 0, dated 16/02/2015 later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/03/2015</td>
<td>Active</td>
<td>10052502</td>
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<tr>
<td>AEROTEC GROUP</td>
<td>S.A.S transferred to 306218 with 70386</td>
<td>EC 135 T2+</td>
<td>Night Vision Goggles (NVG) Compatible Lighting System. Description: This STC adds interior and exterior lighting fo...</td>
<td>The Night Vision Goggles (NVG) Compatible Lighting System. Description: This STC adds interior and exterior lighting fo...</td>
<td>Flight Manual Supplement “ATG-M-EC135-01-04”, dated 09/02/2015 or later revisions of the above listed documents approved by EASA Design Definition File “ATG-DD-EC135-01-02”, dated 02/03/2015-User’s Manual “ATG- DTU-005-001_R03”, dated 02/03/2015-Installation Manual “ATG-DD-EC135-01-02”, dated 02/03/2015.</td>
<td>None</td>
<td>06/03/2015</td>
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<td>STC Holder</td>
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<td>Description</td>
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<td>Limitations/Conditions</td>
<td>System Record</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>LUTTHANSA TECHNIK AG</td>
<td>747-8</td>
<td>Use of INFI in the Cockpit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval.</td>
<td>None</td>
<td></td>
<td>09/02/2015</td>
<td>Active</td>
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<tr>
<td>VHI AEROSPACE Inc.</td>
<td>Airbus Corporate Jet Modification CJ1486 VIP Project “Highland”</td>
<td>The Certification Basis for the original product as amended by this certificate/approval.</td>
<td>The Certification Basis for the original product as amended by this certificate/approval.</td>
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<tr>
<td>CAA UK RESPONSIBILITY</td>
<td>EC 135 T2+</td>
<td>Surveillance Role Equipment installation and NVIS compatible lighting system as per BOND/MOD/103.</td>
<td>The Certification Basis for the original product as amended by this certificate/approval.</td>
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<td>BOND HELICOPTERS EUROPE LTD.</td>
<td>F9XW-D0013-005</td>
<td>Validation of FAA STC STS20973NY Installation of ADS-B Out System on aircraft equipped with Honeywell Primus 2000 avionics system.</td>
<td>The Certification Basis for the original product as amended by this certificate/approval.</td>
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<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>F9XW-E0144-001</td>
<td>- Flight Manual Supplement V81-0614-FMS, Revision 02, dated October 3, 2007 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canada Bilateral Agreement - VIH Aerospace, Inc. Master Drawing List.</td>
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</table>
List of EASA Supplemental Type Certificates

**STC Holder**

10052547
HAI CO CABIN SOLUTIONS, LLC
FAA STC ST04170AT
A320-214
Installation and activation of an EFE System
The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

- Instructions for Continued Airworthiness, TIMCO document No. 13T310L009(3)(c) Revision A, dated 11 December 2014 or later revisions of the above listed documents approved by EASA.
- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10052548
MARSHALL AVIATION SERVICES LIMITED
CAA UK RESPONSIBILITY
B200CGT, B300C, B200, B200C, B200GT
Introduction of Rockwell Collins TCAS-4000 (TTR-4000) with Version 7.1 software as an update to an existing TCAS-4000 with Change 7.0 software in accordance with Marshall Aerospace Services Ltd Modification NoSC0530A.
The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10052551
AVIATION COMMUNICATION ANDREWELLAN CE SYSTEMS, L.L.C. (ACSS)
FAA ST0254LA SEE APPROVED MODEL LIST
Upgrade of ACSS ACAS II Version 7.0 to incorporate EASA AD 2012-02 (various aircraft models are included)
The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

- EASA AML List or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/004/CF or subsequent revisions of this decision and/or the Technical Implementation Procedures of EU USA Bilateral Agreement.
- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10052556
PMV ENGINEERING
767-300, 767-400ER 767-200, 767-300 TCAS 7.1 upgrade on B767 The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

- Master Document List AVI-00750-MDL-R03 dated 5 Mar 2015: Aircraft Flight Manual Supplements as per AVI-00750-AMS-R01 dated Feb 2015: Instructions for Continued Airworthiness as per AVI-00750-ICA-R01 dated 9 Feb 2015 or later revisions of the above listed documents approved by EASA.
- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10052558
E4C TECHNOLOGIES, L.L.C.
FAA STC ST0251LA BOEING 767-300 Installation of seat tracks at LBL and RBL 33.0 The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

- Master Data List MDL2135-2007-10 rev Cor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/004/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU USA Bilateral Agreement.
- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

03/07/2024
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<th>STC Holder</th>
<th>STC No.</th>
<th>Rev.</th>
<th>STC Type</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Ref.</th>
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<tr>
<td>KNISSLEY WELDING, Inc.</td>
<td>TSIO-520-R</td>
<td>1</td>
<td>FAA STC</td>
<td>Exhaust System Modification/installation of a modified exhaust system in accordance with Kniissley/Welding, Inc., Modified Master List (MMML) 1 Revision G, dated May 19, 1993, or later FAA approved revision.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Instructions M-2-1, Rev. B, dated March 6, 1992 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/94/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/EASA Bilateral Agreement</td>
<td>Approval of this change in type design is limited to engines specifically identified above, and does not constitute approval for installation on an aircraft. A separate approval for installation on each aircraft must be obtained prior to installation of this design change. It must be determined that...</td>
<td>11/02/2015</td>
<td>Active</td>
<td>10502561</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>CESSNA FR172J, FR172K</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>Installation of S-TEC system SS/55. Two Axs Automatic flight guidance system, model ST-412, according to Bulletin No. 712 rev. 2 dated28-Feb-2001.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>To be installed according to S-TEC Master Drawing List No. 9207 rev. Bdated 24-Feb-2001(14V system). To be operated in accordance with FAA/DAS approved S-TEC APA-5 PN91223 rev.1 dated 25-Feb-2001. To be maintained i.e.: Instruction for continued Airworthiness 81328rev.2 or later revisions of the above...</td>
<td>As per AFMS Prior to installation of this design change it must be determined that... the...</td>
<td>11/01/2015</td>
<td>Active</td>
<td>10502563</td>
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<tr>
<td>FLYING COLOURS CORP.</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>757-300</td>
<td>Issue 1</td>
<td>Installation stone wash flooring in galley and lavatories</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>RESTRICTED TO ASN 9581</td>
<td>12/03/2015</td>
<td>Active</td>
<td>10502565</td>
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<tr>
<td>PMV ENGINEERING</td>
<td>737-900ER, 737-900ER, 737-700, 737-700, 737-800, 737-300, 737-400, 737-500, 737-600, 737-700, 737-800, 737-900</td>
<td>TSAS 7.1 upgrade on 8737</td>
<td>FAA STC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Master Document List AV-01082-MDL-R01 dated 5 Mar 2015; -Aircraft Installation Instructions per AV-01082-ICA-R00 dated 20-Feb-2015;</td>
<td>TRANSPORT CANADA STC NO. CLSLAS-0011D ISSUE 1 DATED 12 JANUARY 2015-03-11MT</td>
<td>To be maintained i.e.: Instruction for continued Airworthiness 81328rev.2 or later revisions of the above...</td>
<td>Prior to installation of this design change it must be determined that...</td>
<td>12/02/2015</td>
<td>Active</td>
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<td>B/E AEROSPACE, Inc.</td>
<td>757-300, 757-200FF, 757-200</td>
<td>TSAS 7.1</td>
<td>FAA STC</td>
<td>Installation of Vacuum Toilet Assembly Retrofit Kit</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that... the...</td>
<td>12/02/2015</td>
<td>Active</td>
<td>10502568</td>
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<td>Bournemouth Aviation (Consultants) Ltd</td>
<td>B3360, B300C</td>
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<td>CAA UK RESPONSIBIL ITY</td>
<td>B3300, B300C</td>
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<td>Limitations/Conditions</td>
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<td>STC Holder</td>
<td>Status</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>PMV ENGINEERING</td>
<td>Active</td>
<td>757-200, 757-200PF, 757-300</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List AVI-00760-MLR-R02 dated 12 Mar 2015-Aeroplane Flight Manual Supplement as per AVI-00760-APM5-R00 dated 3-Feb 2015-Instructions for Continued Airworthiness as per AVI-00760-ICA-R00 dated 3-Feb 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/02/2015</td>
<td>Active</td>
<td>10502582</td>
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<td>B/E AEROSPACE, Inc. - FSI</td>
<td>Active</td>
<td>777-300</td>
<td>First class furniture trim and finish change</td>
<td>- Master Drawing List, Flight Structures Inc. document No. B3FS303- D01, Revision A, dated 9 February 2015- Instructions for Continued Airworthiness . Flight Structures Inc. document No. B3FS303- D15, Revision A, dated 27 February 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/03/2015</td>
<td>Active</td>
<td>10502627</td>
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<tr>
<td>BINDER MOTORENBAU GmbH</td>
<td>Active</td>
<td>ASW 22 BLE 5R ASW 22 BL, ASW 22 BLE, 27 m Ansteckflügel ASW 22 BLE.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Ergänzung des Flughandbuchs betreffend STC ‘27 m Ansteckflügel für das Segelflugzeug / den Motorsegler ASW 22 BL, ASW 22 BLE, 27 m BLE-50R/Ergänzung des Warteraumhandbuchs betreffend STC ‘27 m Ansteckflügel für das Segelflugzeug / den Motorsegler ASW 22 BL, ASW 22 BLE, 27 m BLE-50R/Allgemeines Info...</td>
<td>TN 8 and/or TN10 for ASW 22 BLE/BL/BLE 5R must be done before</td>
<td>18/03/2015</td>
<td>Active</td>
<td>10502646</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>Rev. 3</td>
<td>A318-112</td>
<td>Installation of a cabin VIP interior Revision 2 Major change for conversion to commercial for dual use/private and commercial) Revision 3 Administrative corrections</td>
<td>Compliance checklist: CCL25-19/04M Rev 2 dated 30 Mar 2020- Suplementary Type Definition Document: STD2025-19/04M Issue 1 Rev 0 dated 31 Mar 2020- APM A-00-19/95-AMF Issue 1 Rev 0 dated 27 Mar. 2020-ICA A-00-19/95/ICA Issue 1 Rev 0 dated 04 Dec 2019-Compliance Checklist Report – Ref. 1360000000, Rev.</td>
<td>Prior to installation of this change/repair it shall be determined that the interrelationship between this change/repair and any other previously installed change/repair shall introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/04/2020</td>
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<td>H4 AEROSPACE (UK) Ltd.</td>
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<td>757-200</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>H4 AEROSPACE MODIFICATION SUMMARY AND DECLARATION OF COMPLIANCE NO.H4A1740 ISSUE 1 DATED 19 MARCH 2015.</td>
<td>Any future modifications must consider the WiFi 2.4GHz and 5GHz RF environment during the design. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/03/2015</td>
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<td>10502672</td>
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<td>STC No</td>
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<td>STC Ref</td>
<td>Model(s)</td>
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<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>10052694</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>REV. 1</td>
<td>A320F-6065T</td>
<td>Installation of LPV/ILS (Localizer PERFORMANCE with Vertical Guidance/ Satellite Based Augmentation System) ILS (Instrument Landing System)cockpit function capabilities in accordance with Airbus Transport International (ATI) Change Approval Sheet ST34MDO036 Issue 2 Enhancements. The modification consists...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product ar...</td>
<td>AFM-3 Attached AID ref. (ST34MDO043_ADD Issue 1), or later revisions of the above listed documents approved by EASA. See manuals listed in annex 8 of ATI Change Approval Sheet ST34MDO036 Issue 25. Also see also Annex 7 of ATI Change Approval Sheet ST34MDO036 Issue 2 if approval of evolution of the operator’s M...</td>
<td>20/06/2016</td>
<td>Active</td>
<td>10052694</td>
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<tr>
<td>10052712</td>
<td>PMV ENGINEERING</td>
<td></td>
<td>TCAS 7.1 upgrade on B747</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Master Document List AVI-00800-MDL.R00 dated 3 March 2015 -Eurocontrol Flight Manual Supplement as per AVI-00800-AMFS-R00 dated 27 February 2015 -Instructions for Continued Airworthiness as per AVI-00800-AA-R00 dated 27 February 2015 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/03/2015</td>
<td>Active</td>
<td>10052712</td>
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<tr>
<td>10052729</td>
<td>STIRLING DYNAMICS GMBH</td>
<td>REV. 3</td>
<td>A109S</td>
<td>Installation of Auxiliary Hotel Bracket Revision 3. Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-RPM Supplement 214154-C-AM-2500-001-024r later revisions of the above listed documents approved by EASA. -Drawing Register (Master Drawing List) 214154-C-DR-2500-001-01-Certification Documents (Master Document List) 214154-C-CL-2500-001-01-Compliance Register (CCL) 214154-C-CL-2500-001-01-Means of Compliance</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2022</td>
<td>Active</td>
<td>10052729</td>
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<tr>
<td>10052742</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>FALCON 7X</td>
<td>Cabin 14 Pax Configuration Replacement of existing cabin single seats by new ones and increase 50% of Pax configuration from 12 to 14.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Dassault Falcon Service Change Approval Sheet DFS7035-CAS Issue A from 24 March 2015 for later revisions of the above listed documents approved by EASA. -Dassault Falcon Service Master Document List DFS-DE-17-TX-0705 Issue A or 24 March 2015 -Falcon 7X serial number 40 only</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/03/2015</td>
<td>Active</td>
<td>10052742</td>
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<td>10052751</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td>2, B200, B200C, B300, B300C</td>
<td>Approval of Installation of a Universal LPV Monitor in accordance with Scandinavian Avionics Modification 2092019 Issue 1 dated 27 February 2015.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation of a Universal LPV Monitor in accordance with Scandinavian Avionics Modification 2092019 Issue 1 dated 27 February 2015, and Engineering Order 2092019-EO Issue 1, dated 03 March 2015. Operate in accordance with Scandinavian Avionics Flight Manual Supplement 2092019-AM A Issue 2 dated 23 February 2015.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/03/2015</td>
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<td>10052751</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAAC STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10052805</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td></td>
<td>A340-313</td>
<td></td>
<td>New Premium Economy Class for A340-300 in the main cabin of the main floor on the aircraft, the change consists of the removal of 10 per cent of the Economy Class Seats between door 2 (and door 3) and the installation of new developed Premium Economy Seats instead. The new seating will have a greater pitch...</td>
<td>Statement of Compliance No 3562 Rev 01 Supplemental Type Definition Document STD205-134/1F Issue 1</td>
<td>According to the design related to this STC. This modification must be installed in conjunction with a) SB AM A340-20-106 Introduction of a three part class divider/installation/Minor Change ref MM 635-141 Fx (+) Fx STC 10010964 rev 3, Installation of Non Electric Floor Path Marking System, together...</td>
<td>12/02/2017</td>
<td>Active</td>
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<td>10052811</td>
<td>RAISBECK ENGINEERING Inc.</td>
<td>FAA STC</td>
<td>107055E-T</td>
<td>LEARJET 60 &amp; 60R</td>
<td>Installation of All Fuselage Stowage Lockers</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>-</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/03/2015</td>
<td>Active</td>
</tr>
<tr>
<td>10052812</td>
<td>VT DRB AVIATION CONSULTANTS, INC.</td>
<td>FAA STC</td>
<td>ST116635C</td>
<td>A320-232</td>
<td>Installation of GT &amp; G5 Galley's, on A320-232 aircraft model</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements C8.25 Amendment 1525.20(a)(b) : 405(b) : 307 (ec) : 38(a)(c)(c) : 413(a)(n) : 325 : 853(a)(d)(a)(h) : 406(a) : 130(c) : 130(a)(b) : 1351(a)(v) : 135(n) : 143(i)...</td>
<td>Data List, ref : DL.1899M00 Revision F, dated 16 Dec 2020</td>
<td>1. This modification is only applicable to A320-232 MSN 43882. The following modification must be concurrently installed with this STC/EASA validation of FAA STC ST0398NY. Installation of a VIP Cabin prior to installation of this design change it must be determined that...</td>
<td>31/03/2015</td>
<td>Active</td>
</tr>
<tr>
<td>10052815</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>FAA STC</td>
<td>ST116635C</td>
<td>A310-300</td>
<td>A310 - Zero G Modification</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements Special Condition (s) CRI D-01, Zero-G cabin installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Lufthansa Technik, Airplane Flight Manual Supplement, A-GO-1510-AFM, Issue 01, dated 04 March 2015; Lufthansa Technik, Instructions for Continued Airworthiness, ref : 1899CT25M00 Revision F, dated 11 Dec 2014 (Galley G1), Instruction for Continued Airworthiness, ref : 1899CT25M03 Revision F, dated 11 Dec 2014 (Galley G2), Weight and Balance data, ref : 1899BPP0520 Rev...</td>
<td>Lufthansa Technik, Airplane Flight Manual Supplement, A-GO-1510-AFM, Section 2 - Limitations/Lufthansa Technik, General Substantiation Data APB-CA-02, rev 0, dated 04 March 2015; for data on Handsprings (chapter 3) and Safety Nets (chapter 4) Prior to installation of this design change it must be determined that...</td>
<td>31/03/2015</td>
<td>Active</td>
</tr>
<tr>
<td>10052821</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td>FAA STC</td>
<td>AS 350 B3 AS 350 B2</td>
<td>Installation of TV Communication Equipment for Bridge Helicopter (Configuration SH) and Shooting Helicopter (Configuration SH)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Rotorcraft Flight Manual Supplement RPM NAV-0506, Rev. for later revisions of the above listed documents approved by EASA Instructions for Continued Airworthiness ICA-NAV 0506, Rev. for 0Configuration Shooting Helicopter 25- NAV0506-00 Configuration Bridge Helicopter 25- NAV0506-01 Master Document List...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/03/2015</td>
<td>Active</td>
<td>10528221</td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>FAA STC/ST</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10052853</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>REV. 3</td>
<td></td>
<td>Interior Outfitting/This change describes the installation of a VIP interior including associated components into a 747-8 &quot;green aircraft&quot;: Rev. 1. This revision is issued to remove the limitation to Aircraft Serial Number 37501. All other deviations between the initial STC and the new configuration...</td>
<td>For Aircraft Serial Number 37075: The change is defined with Revision A of Master Data List document602-10A-01.</td>
<td>Aircraft Flight Manual Supplement document602-10A-01.</td>
<td>Installation of additional or alternative airworthiness requirements</td>
<td>24/11/2023</td>
<td>Active</td>
<td>10052853</td>
</tr>
<tr>
<td>10052854</td>
<td>CESSNA ZURICH CITATION SERVICE CEN/GMBH</td>
<td></td>
<td>390</td>
<td>Compliance to ADS-B Out as per DO390B/Compliance to ADS-B Out was shown against EASA CS-ACNS-ADS-B and FAAAC20-16A as per RTCA/DO390B/Implementation of ADS-B out capability into Rockwell Collins Pro Line 21 Avionics Suite with single or dual Rockwell Collins GPS4000S-dual Rockwell Collins TDR40A Tr...</td>
<td>- This certificate does not cover commercial operation (private usage): This certificate does not cover the &quot;Aeroloft&quot; Overhead Sleeping Compartment installation (EASA STC ST11163SC: Installation of a VIP Cabin Interior/STC Delta Eng - A320 VIP Cabin Interior)</td>
<td>- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effects upon the airworthiness of the product.</td>
<td>02/04/2015</td>
<td>Active</td>
<td>10052854</td>
<td></td>
</tr>
<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>J &amp; C ENGINEERING SERVICES LTD.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>EA 3005C</td>
<td>Wing tip pyrotechnic installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td></td>
<td></td>
<td>07/04/2015</td>
<td>Active</td>
<td>10052869</td>
</tr>
<tr>
<td>PALL AEROPower Corporation</td>
<td>FAA STC ST04168AT</td>
<td>737-600; 700; 800; 900</td>
<td>Control Bleed Air Line Filter Kit</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Pall Aeronpower Drawing No.31630KT Revision A dated 08 September 2013.</td>
<td>The installation applies to the referenced aircraft equipped with CFM International CFM56-7B engines.</td>
<td>08/04/2015</td>
<td>Active</td>
<td>10052876</td>
<td></td>
</tr>
<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>FAA STC ST041944T</td>
<td>757-300</td>
<td>Installation of the Rockwell Collins Aircraft Communication, Addressing and Reporting System (ACARS) with Data-link Communication Function</td>
<td>The Certification Basis for the original product as amended by the following paragraphs is applicable to this certificate/approval: the following paragraph(s) is/are later amendment: CS 25 Amendment 4: 25.27(a), 25.37(h)(1)(c), 25.303, 25.305, 25.307(a), 25.561(c)(2)</td>
<td>Master Drawing List, B/E Aerospace document No. 2-128822-007, Revision B dated 9 December 2014 - Airplane Flight Manual Supplement, B/E Aerospace document No. 2-128822-907, Revision B dated 16 January 2015 - Airplane Flight Manual Supplement, B/E Aerospace document No. 2-128822-909, Revision B.</td>
<td>Airplane Flight Manual Supplement, B/E Aerospace document No.2-128822-808 is only applicable to aircraft that have upgraded their Cockpit Voice Recorder (CVR) to satisfy data message recording requirements. Prior to installation of this design change it must be determined through the relationship below...</td>
<td>09/04/2015</td>
<td>Active</td>
<td>10052886</td>
<td></td>
</tr>
<tr>
<td>EMTEQ CANADA Ltd.</td>
<td>REV. 1</td>
<td>TCCA STC SA14-53</td>
<td>BD-700-1A11 BD-700-1A10 MODIFICATION&gt;(&gt;,&lt;) CABIN EQUIPMENT UPGRADE</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-01 GLASS SCREENS OF MONITORS AND DISPLAYS is applicable to this certificate/approval: the following paragraph(s) is/are later amendment: CS 25 Amendment 4: 25.27(a), 25.37(h)(1)(c), 25.303, 25.305, 25.307(a), 25.561(c)(2)</td>
<td>TRANSPORT CANADA STC NO. RA-14-03 ISSUE 1 DATED 04 JULY 2014</td>
<td></td>
<td></td>
<td>13/04/2015</td>
<td>Active</td>
<td>10052888</td>
</tr>
<tr>
<td>SKYTRAC SYSTEMS Ltd.</td>
<td>REV. 2</td>
<td>TCCA SA11-17</td>
<td>ATR72-212A ATR 72-201-202 ATR72-101-102-201-202 &gt;212 ATR 72-101-102 ATR42-300-320-500 ATR 42-300-320 SkyTec Flight Following / SATCOM System This modification is installing a SATCOM system allowing full duplex/ Allsatellite phone communication, automatic message reporting, two-way messaging, flight tracking, and other data transfers. Revision Textends the applicability to the models ATR 42/500, ATR 72-21...</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>ADIS Master Drawing List, ref.: MOL10078 Revision M dated 26th October 2021&gt;Revision B dated 26th October 2021&gt;Revision A dated 26th October 2021&gt;Revision 0 dated 14th October 2021- Flight Manual Supplement, ref.: FM5100078-1 Revision B dated 26th October 2021- Aircraft Flight Manual Supplement, ref.: FMS10078-1 Revision B dated 26th October 2021- Aircraft Flight Manual Supplement, ref.: FMA10078-1 Revision A dated 12 March 2019- Airworthiness L1</td>
<td>STC no. 212A/2012 via MOD 212A with MOD 212A/New Avionic Suite incorporated (commercial designation &quot;ATR 72-212&quot;) Prior to installation of this design change it must be determined whether the following additional or alternative airworthiness requirements: CRI D-01 GLASS SCREENS OF MONITORS AND DISPLAYS is applicable to this certificate/approval: the following paragraph(s) is/are later amendment: CS 25 Amendment 4: 25.27(a), 25.37(h)(1)(c), 25.303, 25.305, 25.307(a), 25.561(c)(2)</td>
<td>29/05/2024</td>
<td>Active</td>
<td>10052890</td>
<td></td>
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<tr>
<td>FIBERGLAS-TECHNIK RUDOLF LINDNER GMBH &amp; CO. KG</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of additional coloured anti-collision markings.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation instructions Fiberglas-Technik Rudolf Lindner document IA/L01, dated 06 March 2015 Aircraft Flight Manual Supplement Fiberglas-Technik Rudolf Lindnerdocument TM L01 AFM, chapter 2.4.4, dated 06 March 2015 or later revisions of the above listed documents approved by EASA-instructions for...</td>
<td>Prior to installation of this design change it must be determined whether the following additional or alternative airworthiness requirements: CRI D-01 GLASS SCREENS OF MONITORS AND DISPLAYS is applicable to this certificate/approval: the following paragraph(s) is/are later amendment: CS 25 Amendment 4: 25.27(a), 25.37(h)(1)(c), 25.303, 25.305, 25.307(a), 25.561(c)(2)</td>
<td></td>
<td>10/04/2015</td>
<td>Active</td>
<td>10052895</td>
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<tr>
<td>STC Holder</td>
<td>System</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>REV. 1</td>
<td>P206T</td>
<td>Title: STC-AT-074. Provisions for autonomous operations. Descriptions: These provisions include new connectors and structure/data design to ease the removal of the instrument panels, service/installation for brakes, nose-wheel steering, throttle, propeller, carburetor heat, 3 C-DC converter, these small variants.</td>
<td>Applicable to Documents Doc. No. STC-AT-074 Rev. 2, dated 23.10.2020/Maintenance Manual Supplement Doc. No. AMM1-STC-AT-074 Rev. 1.</td>
<td>Only provisions are installed by this STC. Autonomous operation of the aircraft as well as the manned operation where the control system of the autonomous operation system operates the aircraft is not part of this STC. STC-AT-074. Prior to installation of this change/repair it must be determined that the installed system operates the aircraft as well.</td>
<td>02/11/2020</td>
<td>Active</td>
<td>10052906</td>
<td></td>
<td></td>
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<tr>
<td>GFD GmbH</td>
<td>REV. 1</td>
<td>35A</td>
<td>IFF System Lower Antenna Installation Rev. 1: Transfer to GFD GmbH</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
<td>Active</td>
<td>10052903</td>
<td></td>
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<tr>
<td>BOND HELICOPTERS EUROPE Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>MBB-BK 117 D-2</td>
<td>EastAnglia Air Ambulance NVIS installation in accordance with BOND/ MOD/109.</td>
<td>Flight Manual Supplement FMN/109-002 Issue 4 dated 9 April 2015 or later revisions of the above listed documents approved by EASA Design decision DPI/ 109-003 Issue 3 dated 1 April 2015</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/04/2015</td>
<td>Active</td>
<td>10052904</td>
<td></td>
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<tr>
<td>PMV ENGINEERING</td>
<td>747-8</td>
<td>Activation of TrueNorth GMS system onboard. The modification consists in the installation of GMS Base Station (IBS) on fwd RH cargo rack. 206-040-387-105, 206-040-383-105, numbers 206-040-383-101, 206-040-383-102, 206-040-383-103, 206-040-383-104, and double seats only as shown in the cockpit is not permitted.</td>
<td>AFMS. Doc. No. SCA-0030-APM6-R01, Rev. R01, dated 12.02.2015- ICA. Doc. No. SCA-0030-ICA-R00, Rev. R00, dated 20.10.2014 or later revisions of the above listed documents approved by EASA Master Data List. Doc. No. SCA-0030- MDL-R03, Rev. R03, dated 11.04.2015.</td>
<td>In the cockpit, the following restrictions apply: The use of wireless or mobile phones in the cockpit is not permitted. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/04/2015</td>
<td>Active</td>
<td>10052911</td>
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<tr>
<td>ROMTEX AERO S.R.L.</td>
<td>EMB-135BJ</td>
<td>Legacy 650 Passenger Seat Restraint Systems Modification</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AFMS. Doc. No. SCA-0030-APM6-R01, Rev. R01, dated 12.02.2015- ICA. Doc. No. SCA-0030-ICA-R00, Rev. R00, dated 20.10.2014 or later revisions of the above listed documents approved by EASA Master Data List. Doc. No. SCA-0030- MDL-R03, Rev. R03, dated 11.04.2015.</td>
<td>The installation of the new restraint system part numbers 502- BH900GNS and 502- BH902GNS under this STC is limited to rearward facing single and double seats only as shown in the cockpit is not permitted. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/04/2015</td>
<td>Active</td>
<td>10052929</td>
<td></td>
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<tr>
<td>KAMATICS CORPORATION</td>
<td>28A, 206B</td>
<td>Installation of KAMATICS KAflex Main Rotor Drivehat Kit in accordance with drawing SKCP2782</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement for Bell 206A2/206B KAflex Main Rotor Drivehat Installation, Revision N/C, dated 28.06.2019 or later revisions of the above listed documents approved by EASA in accordance with EASA EO Decision 2004/014/CF (or subsequent revisions of this decision) and/or the Technical Impr.</td>
<td>Bell 206 rotorcraft with tail rotor drive shaft part numbers 206-040-383-101, 206-040-383-102, 206-040-383-105, 206-040-383-106, 206-040-383-107 and 206-040-383-107 must have the Kamatics KAflex Inertia Kit, drawing 25072744 Revision C or later, installed. Prior to installation of this design change it must be determined that it...</td>
<td>14/04/2015</td>
<td>Active</td>
<td>10052951</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10052964</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>REV.1</td>
<td>CAAS 13-081/58</td>
<td>757-200</td>
<td>B757-200 Interior Reconfiguration to a full first class layout.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-02 Application of heat release and smoke density requirements. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>H4 Aerospace Flight Manual Supplement.</td>
<td>Restricted to M Regional Certification. CON-104 has been annulled as part of Part P has not been embodied on the aircraft. 12 minute Oxygen Generators remain installed in the Lavatories and Galley areas. The aircraft remains as previously certified for 12 minute operations. The HI Alt system is currently deactived.</td>
<td>16/04/2015</td>
<td>Active</td>
</tr>
<tr>
<td>10052955</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>A319-111</td>
<td>Zonal Drying System Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental Type Definition Document, Lufthansa Technik document No. STD201-14/19M Issue 1, Revision 0, dated 27 March 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/04/2015</td>
<td>Active</td>
<td>10052955</td>
</tr>
<tr>
<td>10052957</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 3</td>
<td>35/ 35A/ 36/ 36A</td>
<td>Upgrade TCAS II to ACAS II Change 7.1 Software.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. ACC Columbia Jet Service, Airplane Flight Manual Supplement LR35-15002-80-MF-1, Rev I, dated 25 March 2015. 2. ACC Columbia Jet Service, Instructions for Continued Airworthiness LR35-15002-850-M40, Rev I, dated 17 March 2015</td>
<td>Only applicable to LR35/36 or LR 35/36A with STC SA0561M-D (LBA validated).</td>
<td>26/01/2022</td>
<td>Active</td>
<td>10052957</td>
</tr>
<tr>
<td>10052959</td>
<td>ROMTEX AERO S.R.L.</td>
<td></td>
<td>B737-800</td>
<td>Passengers safety belts installation.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25 at amendment 14,25.961(b), 25.962, 25.625(b), 25.786(b)(d)(3) (i), 853(a), 1301 (a),1309(a)(2), 1529: The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List 02-400003-MDL-001 rev 00. Component Maintenance Manual supplement 02-400003-CMMIS-001 rev 00. Latest revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/04/2015</td>
<td>Active</td>
<td>10052959</td>
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<tr>
<td>10052960</td>
<td>VECTOR AEROSPACE HELICOPTER SERVICES INC.</td>
<td>FAA STC SR20415AK</td>
<td>AS 332 C, AS 332 L, AS 332 L1</td>
<td>Installation of Cockpit Voice and Data Recorder System in accordance with Vector Aerospace Helicopter Services USA, Inc. Master Drawing List MDL HP 332-23-71-102. Rev G, dated 29/01/2015</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement FMS HP 332-23-71-301, Rev A, dated 04/07/2014</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/04/2015</td>
<td>Active</td>
<td>10052960</td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NA</td>
<td>STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10052962</td>
<td>CEIKA LETECKA SERVISNI A.S.</td>
<td>0</td>
<td>L-490/VP-E</td>
<td></td>
<td></td>
<td>Installation of the LPV monitor in order to comply with the MC20-28 for LPV operation.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CRI A-01, Issue 1, dated 15.05.2014.The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Document List, AC 248 - 0017, rev. 1. OCOMPLIANCE CHECKLIST AC 248 - 6101, rev. 0. FLIGHT MANUAL SUPPLEMENT FLIGHT MANAGEMENT SYSTEM (FMS) ULS-1FA x LPV, AC232-2001-1 (n) dated 15.07.2015 - rev. 0.1 later revisions of the above listed documents approved by EASA.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 2.A.109 or 21.A.451(b) (as applicable).</td>
<td>16/04/2015</td>
</tr>
<tr>
<td>10052971</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>0</td>
<td>FAA STC ST0198AT</td>
<td>MD-11, MD-11F</td>
<td></td>
<td>Installation of the Honeywell RTA-4A Weather Radar Receiver/Transmitter in the Boeing MD-11 series</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>699-50125, Revision B (1999 Jan 2007) - Master Drawing List (MDL) 795-0077-150, Revision - (2014 Sept 25) - EASA Certification Basis/Comparison Reporter later revisions of the above listed documents approved by EASA in accordance with EASA EDDocument 2004/014/CF (or subsequent revisions of this decision)</td>
<td>None</td>
<td>16/04/2015</td>
</tr>
<tr>
<td>10052976</td>
<td>WHELEN ENGINEERING COMPANY Inc.</td>
<td>0</td>
<td>FAA STC SA0195EA</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td></td>
<td>Installation of Whelen Anti-Collision Strobe Light Systems Part Number: 01-070000(-1), 01-070020(-2), 01-070022(-3), 01-077060(-1), 01-077060(-2), 01-077060(-3), 01-077060(-4), 01-077100(-1), 01-077100(-2) (as applicable)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation and Maintenance according to Whelen Anti-Collision Light Systems Installation and Service Manual Document 06131 Rev. B (June 2010). or later revisions of the above listed documents approved by EASA in accordance with EASA EDDocument 2004/014/CF (or subsequent revisions of this decision)</td>
<td>1. These lights comply with the anti-collision light standards of the FARs as follows: - The red or combined red/white lens; those effective on or prior to August 10, 1971. - The white lens; those effective on August 11, 1971.2. Install the following lenses (as applicable).</td>
<td>17/04/2015</td>
</tr>
<tr>
<td>10052988</td>
<td>BERINGER AERO</td>
<td>REV. 3</td>
<td>LAK-17B FES LAK-17A, LAK-17AT, LAK-17F FES LAK-17A</td>
<td></td>
<td></td>
<td>Installation of main wheels and brake system P/N: RF-002(B) and EA-006Rev. 1: Update of part numbers including the newwheel P/N: RF-019(B)(which supercedes the P/N: RF-002(B)(Rev. 2: Update with new servicing instructions within the CAA document and addition of SensAR device as an optionRev. 3: Update...</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification file: CR-MOD001-STC-011 REV 0 dated 30 Nov. 2023</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>01/03/2024</td>
</tr>
<tr>
<td>10053008</td>
<td>GARMIN INTERNATIONAL Inc.</td>
<td>REV. 2</td>
<td>FAA STC SA02153A-D</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td></td>
<td>Installation of Garmin GI600 Primary Flight Display (PFD) and multifunction Display (MFD) system.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC.There are the requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>1. Master Drawing List, GI600 PFD MFD System AML STC, 005-C031.5-00, Revision D, dated 1 February 2011; 2. Airplane Flight Manual Supplement, GI600 PFD/ MFD System AML STC, 190-00401-01, Revision E dated 28 January 2011; 3. GI600 PFD/ MFD System Instructions for Continued Airworthiness, 190-00301-00, Revision 4...</td>
<td>1. Aircraft equipped with Garmin GI600/G1000 Integrated Cockpit System are not eligible for this STC. 2. Aircraft equipped with Rockwell Collins ProLine 21 IntegratedAvionics System are not eligible for this STC. Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>04/10/2018</td>
</tr>
<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10053014</td>
<td>THE MONROE COMPANY, LLC</td>
<td>REV 1</td>
<td>FAA STC STO9611CH</td>
<td>CESSNA 550</td>
<td>CESSNA 550</td>
<td>Installation of Dual Iridium/GPS Antenna for the original product remains applicable to this certification approval.</td>
<td>Aircraft Flight Manual Supplement document ref. AE14251IFMdated release dated 02 April 2015.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/04/2015</td>
<td>Active</td>
</tr>
<tr>
<td>10053025</td>
<td>KAMATICS CORPORATION</td>
<td>FAA STC</td>
<td>BELL 206A, 206 B</td>
<td>BELL 206A</td>
<td>Installation of KAMATICS Rotor Brake kit in accordance withdrawing SKCP2075</td>
<td>Flight Manual Supplement for Bell 206A/206B Mx ref. motor brake modification Revision 2, dated September 30, 1994.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/04/2015</td>
<td>Active</td>
<td>10053025</td>
</tr>
<tr>
<td>10053031</td>
<td>LMI AEROSPACE, INC DSA TAES INC.</td>
<td>FAA STC</td>
<td>767-300</td>
<td>767-300</td>
<td>Installation of an In-Flight Entertainment System Power Control Switch</td>
<td>Aircraft Flight Manual Supplement document ref. AE14251IFM dated release dated 15 April 2015.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/04/2015</td>
<td>Active</td>
<td>10053031</td>
</tr>
<tr>
<td>10053036</td>
<td>AEROTECH &amp; CONCEPT</td>
<td>A310-304</td>
<td>Hangar</td>
<td>CRJ-200</td>
<td>CRJ-200</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval.</td>
<td>As per paragraph 5.2 “Limitations” of Aerotec document ref. AE14251PCO, Issue 3.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/04/2015</td>
<td>Active</td>
</tr>
<tr>
<td>10053040</td>
<td>AEROTECH &amp; CONCEPT</td>
<td>B767-300</td>
<td>Installation of Dual Iridium/GPS Antenna</td>
<td>Boeing 767</td>
<td>Boeing 767</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval.</td>
<td>Master Data List, Aerotec document ref. AE14251DMC, Revision 1, dated 21 April 2015.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/04/2015</td>
<td>Active</td>
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<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>CENTEX AEROSPACE, INC.</td>
<td></td>
<td>FAA STC SA1142SC</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Construction and Installation of CanTax Aerospace Inc; Auxiliary fuel saddle tanks</td>
<td>The Certification Basis for the original product as amended by the following: Additional or alternative airworthiness requirements: The original aircraft certification basis as defined in the TCDs is satisfied following FAR Part 23 sections (Amendment 23-56) that apply to the saddle tank installation approved...</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/04/2015</td>
<td>Active</td>
<td>10053071</td>
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<tr>
<td>Diao Aviation A/S</td>
<td></td>
<td>BN2B-27, BN2B-21, BN2B-26, BN2A-9, BN2A-20, BN2A-21, BN2A-26</td>
<td>Weather Radar and Avidyne EX-500 Multi-Function display in accordance with FAA-0543</td>
<td>The Certification Basis for the original product as amended by the following: Additional or alternative airworthiness requirements: The initial TCDs is satisfied following FAA-0543 that apply to the radar and multi-function display installation approved...</td>
<td>Flight Manual Supplement FAA-0543-ECO-02</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/04/2015</td>
<td>Active</td>
<td>10053072</td>
<td></td>
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<tr>
<td>TASS-EU Ltd</td>
<td></td>
<td>DHC-8-202</td>
<td>Upgrade of TCA8 7.0 to 7.1 on DHC-8-202</td>
<td>The Certification Basis for the original product as amended by the following: Additional or alternative airworthiness requirements: The initial TCDs is satisfied following FAA-0543 that apply to the multi-function display installation approved...</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/04/2015</td>
<td>Active</td>
<td>10053088</td>
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<tr>
<td>AERODATA AG</td>
<td></td>
<td>EMZ TA 0523</td>
<td>FALCON 10</td>
<td>The Certification Basis for the original product as amended by the following: Additional or alternative airworthiness requirements: The initial TCDs is satisfied following FAA-0543 that apply to the multi-function display installation approved...</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/04/2015</td>
<td>Active</td>
<td>10053097</td>
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<tr>
<td>AAXICO TECHNICAL SERVICES INC.</td>
<td>REV. 1</td>
<td>FAA STC 04177AT</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>LED Sealed Beam Replacement Lamps. for various Part 25 Airplanes. Installation of a Talon Aerospace LED Replacement General Purpose Lamps Rev 1. Transfer of STC</td>
<td></td>
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<td></td>
<td>29/06/2021</td>
<td>Active</td>
<td>10053101</td>
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<td>STC Holders</td>
<td>Rev.</td>
<td>Foreign/NA AIC Std.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record No.</td>
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<td>AIRWOLF AEROSPACE L.L.C.</td>
<td>10053107</td>
<td>REV. 1 FAA STC</td>
<td>SRA0308CCH</td>
<td>212 205A-1</td>
<td>Replacement of Tension-Tension Strip Assy/Revision 1: Addition of new Airwolf tension-tension strip part numbers AA-204-310-101-101C and AA-204-310-101-10C as an option to the existing AA-204-310-101-101 and AA-204-310-101-103 parts.</td>
<td>The Certification Basis for the original product remains applicable to this certificate's approval, except where amended by additional or later amendments if indicated on FAA STC SRA0308CCH. The requirements for environmental protection and the associated certified noise and/or emissions levels of the on.</td>
<td>27/11/2020</td>
<td>Active</td>
<td>10053107</td>
<td></td>
</tr>
<tr>
<td>DELTA Q DESIGN Inc.</td>
<td>10053108</td>
<td>REV. 1 FAA STC</td>
<td>ST1112015C</td>
<td>737-900ER 737-900 737-800/900-900ER 737-700 737-600 737-400 737-300 400-500 500-700 300 737-300-400 10053131</td>
<td>Structural modification necessary for the installation of an ENSAMIT-380 SATCOM Antenna Revision 1: three additional installation locations for the Honeywell/Avionics AMT-380 SATCOM Antenna consisting of one (1) new location on each 737-800 and/or -900ER and two (2) new locations on the 737-800ER aircraft...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.371(a)(6) of 435.253(a)(6), 25.301(a)(6), 25.303, 25.305(a)(6), 25.307(a)(6), 25.31(a)(6) 25.33, 25.39, 25.63...</td>
<td>26/04/2022</td>
<td>Active</td>
<td>10053108</td>
<td></td>
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<tr>
<td>SCANINAVIAN AVIONICS DESIGN AS</td>
<td>10053125</td>
<td>2. BEECH 200C200C 1. BEECH200/200C</td>
<td>Installation of Dual Universal EFI-890R system interfaced with DualGarmin GTN-750 as primary navigation source (SA0 Project No. 2086597)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 23.371 of 434.371 of 34-FAR 23.341 of 34-FAR 23.581 of 34.371 The requirements for environmental protection and the associated identified noise...</td>
<td>In case an UNS-IL FMS is installed on the aircraft in parallel with this STC (SAO No. 2086597), it could be used exclusively for special mission (e.g. Flight inspection) only. For commercial operation flight plan of the FMS shall not loaded. The installer must determine whether this design change is comp...</td>
<td>28/04/2015</td>
<td>Active</td>
<td>10053125</td>
<td></td>
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<tr>
<td>SAA SOLUTIONS FOR AVIATION, S.A.</td>
<td>10053131</td>
<td>REV. 1</td>
<td>BN2A-20, BN2A-21</td>
<td>Installation of an MX15D camera outside the fuselage and installation/adaptation of an existing passenger seat/Rev. 1: Addition of model BN2A-20</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 23.371 of 434.371 of 34-FAR 23.341 of 34-FAR 23.581 of 34.371 The requirements for environmental protection and the associated identified noise...</td>
<td>The BN001-RPT01 Master Drawing List Issue 4, date 19/03/2009 No. 13,001 Modification Description Date 3, date 18/03/2009 No. 13,001 Flight Manual Supplement Issue 2, date 18/03/2009 No. 13,001 Maintenance Manual Supplement Issue 2, date 18/03/2009 additional revisions of the above listed document (x) approved/...</td>
<td>30/03/2020</td>
<td>Active</td>
<td>10053131</td>
<td></td>
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<tr>
<td>WIPARE Inc.</td>
<td>10053138</td>
<td>REV. 2 FAA STC</td>
<td>SA067ICG AND SA149ICG</td>
<td>208</td>
<td>Installation of vortex generators Optional installation of Pratt &amp; Whitney PT6A-140 engine in accordance with EASA STC 10653138 (875 SHP) Optional installation of Pratt &amp; Whitney PT6A-424 engine in accordance with EASA STC 10654107 (850 SHP) Optional installation of Pratt &amp; Whitney PT6A-140 engine max...</td>
<td>The Certification Basis for the original product remains applicable to this certificate's approval, except where amended by additional or later amendments if indicated on FAA STC SA070ICG. This approval involves a change with an impact on the noise database. The environmental protection requirements are a...</td>
<td>Aircraft incorporating this STC are authorized a maximum takeoff gross weight of 8360 lbs.AIR SPEED LIMITS: V (Manoeuvring) at 8360 lbs. = 153 KIAS MAXIMUM WEIGHT: 8360 lbs. Takeoff and landing weight 8360 lbs. Landing 8360 lbs...</td>
<td>20/07/2021</td>
<td>Active</td>
<td>10053138</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10053146</td>
<td>O.G.M. DESIGN GmbH</td>
<td></td>
<td>10053146</td>
<td>BD-700-1A10</td>
<td>Cabin Management System Replacement - Installation of Honeywell Ovation II Select</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>OCM Certification Report No. BBD7-OCM23-001 Revision 0 dated 28 April 2015</td>
<td>Restricted to ASN9084</td>
<td>20/04/2015</td>
<td>Active</td>
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<tr>
<td>10053162</td>
<td>AEROTEC &amp; CONCEPT</td>
<td></td>
<td>757-200</td>
<td>Cabin Management System Replacement</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, Aerotec document No. AE1426MDL, Revision 0, dated 20 April 2015 - Aerospace Flight Manual Supplement, Aerotec document No. 14267AFMS(-&gt;&lt;&gt;) Revision 0, dated 23 February 2015 - Instruction for Continued Airworthiness, Aerotec document No. AE14267CA(-&gt;&lt;&gt;) Revision 0, dated 23 February 2015</td>
<td>Prior to installation of this design change it must be determined that the interdisciplinary relationship between this design change and any other previously installed design change(s) or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/06/2015</td>
<td>Active</td>
<td>10053162</td>
</tr>
<tr>
<td>10053203</td>
<td>AVONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>C. 441</td>
<td>404 F406 A. 421B, 421C, 425, 404, 441 A. 414, 414A, 421, 421A, 429 A. 402B, 402C, 411, 411A A. 401, 401A, 410B, 402, 402A</td>
<td>Single/dual installation of Garmin G600 System(s) including SVT</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 326, 351, 303, 305, 307, 561, 575, 603, 605, 611, 613, 671, 777, 773, 777, 863, 867, 1182, 1301, 1303, 1309, 1311, 1321, 1322, 1323, 1326, 1327, 13...</td>
<td>Flight Manual Supplement EASA-2011-036-FMS-05-00, Rev. 06 later revisions of the above listed documents approved by EASA. Project Description EASA-2011-036-PRD-04-00, Rev. 06-Instructions for Installation EASA-2011-036-IR-10-00, Rev. 00-Instructions for Continued Airworthiness EASA-2011-036-ICA-11-00...</td>
<td>This STC is not applicable to aircraft equipped with 50 A alternators. For automatic AHRS monitoring a suitable Garmin G500 must be installed and be available for connection to the G600 system.</td>
<td>04/06/2015</td>
<td>Active</td>
<td>10053203</td>
</tr>
<tr>
<td>10053214</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>REV. 1</td>
<td>45(LEARJET 45)</td>
<td>TCAS TPU07A to B Upgrade to Version 7.1 (known) Upgrade of the existing TPU67A TCAS Processor from Version 7.0 to TPU67B/Version 7.1 by upgrade via Honeywell SB TPU 67A-34/28-37/67B upgradement by a TPU67B.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Flight manual Supplement, AG994-071, rev Altimetrations for Continued Airworthiness, AG994-065, rev. FR, dated 01 April 2015 or later revisions of the above listed documents approved by EASA. Project Description EASA-2011-036-PRD-04-00, Rev. 06-Instructions for Installation EASA-2011-036-IR-10-00, Rev. 00-Instructions for Continued Airworthiness EASA-2011-036-ICA-11-00...</td>
<td>Prior to installation of this design change it must be determined that the interdisciplinary relationship between this design change and any other previously installed design change(s) or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/06/2015</td>
<td>Active</td>
<td>10053214</td>
</tr>
<tr>
<td>10053225</td>
<td>TC INTER-INFORMATICS A S</td>
<td>REV. 1</td>
<td>A330-302, A330-303, A330-323, A330-32F, A330-243F, A330-201, A330-203, A330-223</td>
<td>A330 Improved Air Filter Element installation This modification is to substitute to the originally installed airrecirculation filter elements by improved air filter elements EO/01-001-1(-&gt;&lt;&gt;) without modifying the system original design and associated maintenance procedures.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 14 -25.01 ; 401 ; 601 ; 605 ; 611 ; 613 ; 631 ; 853 ; 1301 ; 1309 ; 1529 ; 1541...</td>
<td>Definition Document, ref: DB1-D01 Revision 1, dated 6 May 2015 - Instructions for Continued Airworthiness, ref: ICA-14-007, dated 26 March 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>18/06/2015</td>
<td>Active</td>
<td>10053225</td>
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<tr>
<td>10053238</td>
<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A321-231/232 A321-211/212-213 A321-111/112-131 A320-231/232-233 A320-211/212-214/215-216 A319-131/1-132/133 A319-111/112-113/114-115</td>
<td>Electronic Flight Bag (EFB) Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Tenencia Ltd Report TEN 768/7 Issue 2 Flight Manual Supplement later revisions of the above listed documents approved by EASA. Tenencia Ltd Modification TEN 768 Issue 2 and all its referenced documents</td>
<td>Prior to installation of this design change it must be determined that the interdisciplinary relationship between this design change and any other previously installed design change(s) or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/06/2015</td>
<td>Active</td>
<td>10053238</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10053241</td>
<td>SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT HELICOPTERS DATAMANAGEMENT SOLUTIONS &amp; SERVICES</td>
<td>REV. 1</td>
<td>429</td>
<td>Installation of Flight Data Monitoring and Transmitting System Helionix/2 JUM or Helionix/2+ NUM/Rev 1 STC holder name change from I.S.E.I. International Services to SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT - HELICOPTERS DATA MANAGEMENT SOLUTIONS &amp; SERVICES.</td>
<td>The Certification Basis for the original product as amended by following additional or alternative airworthiness requirements: Special Condition (c) CRI F-01 Lithium Batteries, closed at issue 3 on/18/03/2015. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended by this certificate/ approval.</td>
<td>Rotorcraft Flight Manual Supplement, doc. ref. N11-BELL429 HV2 HV2+NUM- E-B-02 Rev.03 dated 24/04/2015(+). Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce aadverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce aadverse effect upon the airworthiness of the product.</td>
<td>21/02/2020</td>
<td>Active</td>
<td>10053241</td>
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<tr>
<td>10053243</td>
<td>KAMAN CORPORATION RCF WHEEL AND BRAKE, LLC</td>
<td>FAA STC SA1077QL</td>
<td>2) C90A, C90GT, C90GTI 1) 70, B90, C90, E90, H90 1) 65, A65 1) 65-85, 65-90, 65-A60 1) 65-80, 65-A80, 65-B80</td>
<td>Replacement of existing Nose Wheel with Cleveland Nose Wheel Part No48-204.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended by this certificate/ approval.</td>
<td>In accordance with Parker Harwin Conversion Kit 199-126, Revision K, dated December 4, 2008 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures applicable to this certificate/ approval.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce aadverse effect upon the airworthiness of the product.</td>
<td>07/05/2015</td>
<td>Active</td>
<td>10053243</td>
</tr>
<tr>
<td>10053269</td>
<td>AERODATA AG</td>
<td>FAA STC A109E</td>
<td>B300</td>
<td>Installation of AD-AFIS-0116 into B300</td>
<td>The Certification Basis for the original product as amended by following additional or alternative airworthiness requirements. The following CS-23 Amr. 3 paragraph apply:23.1, 23.3(d), 23.20(a)(2), 23.25, 23.251, 23.301(a) and (c), 23.303, 23.305, 23.307, 23.337(a) and (b), 23.34(a) and (c), 23.3...</td>
<td>Installation, Operation and Maintenance according to the Master Drawing List document number MDL-At C-856 Rev. 3 dated 7th May 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>11/05/2015</td>
<td>Active</td>
<td>10053269</td>
<td></td>
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<tr>
<td>10053271</td>
<td>AGUSTAWESTLAN PHILADELPHIA, INC</td>
<td>FAA STC A1108</td>
<td>A108E</td>
<td>Installation of Enhanced Ground Proximity Warning System (EGPWS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended by this certificate/ approval.</td>
<td>Flight Manual Supplement AAC-EMG-03-006, Rev. B or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement. Instructions for Continuous Airworthiness Maintenance will...</td>
<td>None</td>
<td>11/05/2015</td>
<td>Active</td>
<td>10053271</td>
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<tr>
<td>10053276</td>
<td>FIELD AVIATION EAST, LTD.</td>
<td>REV. 2</td>
<td>TCCA O-LSA14-143D</td>
<td>DHC-8-202</td>
<td>Universal UNS-1(E) Flight Management System Upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended by this certificate/ approval.</td>
<td>- Universal UNS-1(E) Flight Management Systems are approved for IFR/En-route Terminal and Non-Precision approach operations only (LPV and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement. Instructions for...</td>
<td>25/08/2015</td>
<td>Active</td>
<td>10053276</td>
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<td>EASA STC no</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10053278</td>
<td>AIRCRAFT DEVELOPMENT AND SYSTEMS ENGINEERING (ADSIE) B.V.</td>
<td>MYSTERE-FALCON 900</td>
<td>Installation of AeroGlass Lens for window panels and AeroGlass Mirror fotobrery finished in Falcon 900 aircraft. The design change consists in replacing all protective inner windowpanes in the passenger cabin and cabinet doors with mirrors manufactured by Aviation Glass &amp; Technology terminated.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 14 for all affected requirements. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>ADSE Design Definition List document ref. 14-DO-014 Rev 3 dated 03.05.2015 ADSE Design and Certification Plan ref. 14-DO-012 Rev 2 dated 09.05.2014 ADSE Compliance Check List ref. 14-DO-015 Rev 5 dated 08.05.2015 for later revisions of the above listed documents approved by EASA.</td>
<td>SIN 175</td>
<td>11/05/2015</td>
<td>Active</td>
<td>10053278</td>
<td></td>
</tr>
<tr>
<td>10053282</td>
<td>SABENA TECHNICS B.C.C.</td>
<td>REV. 4</td>
<td>737-8/9 737-600-700-800-900-900ER</td>
<td>Mechanical &amp; electrical installation for personal tablet computer. Revision 1: correction of project scope; only one docking station size. Revision 2: the configuration with the mechanical installation only without the electrical provisions, can be installed. Revision 3: A new size of tablet docking-sta...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: Applicable regulations. CS 25 Amendment 4 (refer to the associated Certification Plan) dated 11-05-1053 and R6, sections 5 and 8.</td>
<td>Master Data List, ref. 0659-01-A.2-001.R6. Instructions for Continued Airworthiness (Supplement), ref. 0659-01-A.2-111.R6-APM.</td>
<td>21/06/2019</td>
<td>Active</td>
<td>10053282</td>
<td></td>
</tr>
<tr>
<td>10053283</td>
<td>NECASER AVIATION GROUP S.p.A.</td>
<td>REV. 1</td>
<td>629</td>
<td>Mango VIP interior PIN 6842000001-X001 consisting of a cabin interior VIP customization in terms of furnishing, seats, key-ports and entertainment system. This STC revision 01 is issued under the Life Vest Provision Installation PIN 6842000001-XXX.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>6B429RPM-1-1 6B429 VIP Interior Rotorcraft Flight Manual Supplement, Revision 02 dated June 2016, ICAOALS as per Chapter 466 of 6B429NIM-2-1 6B429 VIP Interior Owner (Manual), Revision 04(&lt;; v&gt;) dated June 2015. For later revisions of the above listed documents approved by EASA 6B420000N-004 Rev. G, 6B49 V.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/06/2016</td>
<td>Active</td>
<td>10053283</td>
</tr>
<tr>
<td>10053287</td>
<td>ALPINE AEROTECH LP</td>
<td>212, 412, 412EP</td>
<td>Installation of C Box Drain Reservoir. This Reservoir replaces the original manifold and prevents the drain oil from immediately venting overboard.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation Instructions (ICA) AAL-051-500-011, Rev. Ni dated 09.05.2001; Drawing AAL-051-500-011, Rev. Ni dated 08.05.2001 (Configuration) Compliance Report 1602-1, Issue Ni dated 22.10.2001. Maintenance Manual Supplement AAL-051-500-011-MMS, Rev. Ni dated 08.05.2011-15-0123 C-Box Drain reser...</td>
<td>This approval is limited to Bell helmet models 212, 412 and 412EP (the model 412CF is not currently EASA certified).</td>
<td>12/06/2015</td>
<td>Active</td>
<td>10053287</td>
<td></td>
</tr>
<tr>
<td>10053288</td>
<td>AAIICO TECHNICAL SERVICES INC.</td>
<td>REV. 1</td>
<td>FAA ST0400AT</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Titan Aerospace Master Data List MDL-002 Revision 3 and all referenced documents: Titan Aerospace EASA Certification Plan EAS-228-B Revision 01 later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of E.U. USA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/06/2021</td>
<td>Active</td>
<td>10053288</td>
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<tr>
<td>10053293</td>
<td>ELISEN &amp; ASSOCIATES INC.</td>
<td>TCCA STC A-LSA13-015D</td>
<td>CL-600-2B19</td>
<td>Structural Mods &amp; Avionics Relocation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Elisen and Associates Inc Master Data List MDL-378R-757 revision E dated 23 December 2013.</td>
<td>Restricted to ASN 7644</td>
<td>13/06/2015</td>
<td>Active</td>
<td>10053293</td>
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<td>STC Holder</td>
<td>Active</td>
<td>BRTO STCs</td>
<td>Rev.</td>
<td>Foreign/NAA (ICN)</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>ELISEN &amp; ASSOCIES INC.</td>
<td></td>
<td>STC-1647</td>
<td>REV.</td>
<td>TCGA STO-A-LSA13-0140</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Interior Reconfiguration Serial No 7644</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CRI D-01 SIDE FACING DIVIANES: The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unarchived.</td>
<td>Flight Manual Supplement FMS-242D 176 revision Edaled 21 October 2013 later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canada Bilateral Agreement. Edaran Aeronautics Inc. Master List Date MBL-2942.</td>
<td>13/05/2015</td>
<td>Active</td>
</tr>
<tr>
<td>BUCHER LEICHTBAU AG</td>
<td></td>
<td>STC-1647</td>
<td>REV.</td>
<td>MBB-BK117 D-3M</td>
<td>MBB-BK117 D-2 MBB-BK 117 D-2</td>
<td>Implementation of EC 145-AC72-PCI, Quick Release Hook Assy, consists of: a hook, which acts as a securing device for the host operator (ropes down operator a quick release, which attaches the assy to the seat back rail on the rotorcraft Revision 1 Extension of applicability to MBB-BK117 D-...</td>
<td>Flight Manual Supplement FMS-AC72, Revision A024-01-15 &quot;Master Document List/Revision 1 Certification Compliance Report/ Major Change 0013300, Revision ECC/Maxwell Document List 0013301, Revision NC/Restrictions for Continued Airworthiness/ICA-AC72, Revision 01/Flight Manual Supplement FMS-AC72, Rev...</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/12/2023</td>
<td>Active</td>
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<tr>
<td>IKhANA GROUP, LLC</td>
<td></td>
<td>STC-1647</td>
<td>REV.</td>
<td>FAA SA0087LA</td>
<td>B) DHC-6 SERIES 400 A) DHC-6 SERIES 300/310/320 A) DHC-6 SERIES 200/210 A) DHC-6 SERIES 1/001/10</td>
<td>Installation of Pilot Heater Fall LightRev. 1: Name Change from &quot;Ikhana Group Inc.&quot; to &quot;Ikhana Group LLC.&quot;</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment FAR 23, Amend. 1 through 10, 23.1431(a) (d), 23.1307(f), 23.1326, 23.1311, 23.1307, 23.1307, 23.1307, 23.1307...</td>
<td>R. W. Martin, Inc. Master Drawing List PPR-1000, revision B. dated November 18 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/LA Bilateral Agreement.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/06/2021</td>
</tr>
<tr>
<td>IKhANA GROUP, LLC</td>
<td></td>
<td>STC-1647</td>
<td>REV.</td>
<td>FAA STC SA00160LA</td>
<td>DHC-6 SERIES 300 DHC-6 SERIES 200 DHC-6 SERIES 100</td>
<td>Installation of IKhANA Aircraft Services wing box assembly Rev. 1: Name Change from &quot;Ikhana Group LLC.&quot; to &quot;Ikhana Group LLC.&quot;</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unarchived and remain applicable to this certificate/ approval.</td>
<td>R. W. Martin, Inc. Master Drawing List WRR-1000, Issue B. dated December 6, 2007 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/LA Bilateral Agreement.</td>
<td>09/06/2021</td>
<td>Active</td>
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<tr>
<td>CESKA LETECKA SERVISNI A.S.</td>
<td></td>
<td>STC-1647</td>
<td>REV.</td>
<td>10053234</td>
<td>206L-3, 206L-4, 407 206A, 206B, 206L, 206L-1</td>
<td>LASER illuminator provisions. The LASER illuminator is electrically disabled. LASER operation is not EASA approved. By enabling the LASER, the aircraft would transfer to an EASA unapproved configuration. A Key Lock Switch is installed to ensure that the LASER illuminator is not activated without pilot pe...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product unarchived and remain applicable to this certificate/ approval.</td>
<td>Rotorcraft Flight Manual Supplement HC122-3500 Revision 0 Instruction for Continued Airworthiness Laser Illuminator Provision/HC122-2700 Revision 0: for lat revisions of the above listed documents approved by EASA Engineering Order and Certification Program for activation of LASERIlluminator LS-1...</td>
<td>18/06/2015</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>KNI AVIATION ENGINEERING LIMITED</td>
<td>CA-106 RESPONSIBIL ITY</td>
<td>A330-200</td>
<td>Modification of Economy Class Seats</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>KNI-aviation Engineering Limited Master Document List No.140-100-MDL-001-5-R Revision 5 dated 18 May 2015</td>
<td>Restricted to ARN 33K/EASA</td>
<td>19/05/2015</td>
<td>Active</td>
<td>10553340</td>
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<tr>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 1</td>
<td>A330-200</td>
<td>Aer Lingus A330-200 Cabin Modification Rev. 1: Transfer of STC from SR Technics Switzerland to groWNGengineering Partners GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Flight Manual Supplement S.OAED.000377-AFM Rev.0Instruction for Continued Airworthiness S.OAED.000377-ICA Rev0Master Document List S.OAED.000377-MDL-Rev0</td>
<td>EASA STC No. (Ref. EASA PN 0010034239 - KNI seat modification installation) is a prerequisite for the installation and compliance for STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design/changes.</td>
<td>20/04/2021</td>
<td>Active</td>
<td>10553355</td>
<td></td>
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<tr>
<td>KAMATICS CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC S4H68335W</td>
<td>BELL 206L-1, BELL 206L-3</td>
<td>Installation of KAMATICS KAMax Main Rotor drive shaft KIT 59C/3502-3Installation of KAMATICS KAMax Main Rotor drive shaft KIT 59C/3502-3 (with rotor brake)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement dated 13.07.1994 or later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>Rotorcraft Flight Manual Supplement dated 13.07.1994 (or later FAA approved version) is required when SKCP3502-3 KAMax/Max Alphabet rotor brake modification is installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change/changes is required when SKCP3502-3 KAMax/Max Alphabet rotor brake modification is installed.</td>
<td>30/06/2015</td>
<td>Active</td>
<td>10553360</td>
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<tr>
<td>SAFRAN ELECTRONICS &amp; DEFENSEBUSINESS SS UNIT - HELICOPTERS DATAMANAGEMENT NT SOLUTIONS &amp; SERVICES</td>
<td>REV. 1</td>
<td>427</td>
<td>Installation of Flight Data Monitoring System HELICOM V2 NUM and V2+ NUM/Installation of a system for tracking a set of flight and systemparameters during operation of the rotorcraft Rev. 1: STC holder name change from I.S.E.I International Services toSAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT - HELI..</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI F-015C Lithium Battery. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged as...</td>
<td>Rotorcraft Flight Manual Supplement dated 13.07.1994(or later FAA approved version) is required when SKCP3502-3 KAMax/Max Alphabet rotor brake modification is installed. Prior to installation of this design change/changes is required when SKCP3502-3 KAMax/Max Alphabet rotor brake modification is installed.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change/changes is required when SKCP3502-3 KAMax/Max Alphabet rotor brake modification is installed.</td>
<td>27/02/2020</td>
<td>Active</td>
<td>10553361</td>
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<td>STC Holder</td>
<td>STC Ref.</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Interchangeability</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>STC Holder</td>
<td>STC Ref.</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Interchangeability</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>SAAB AB</td>
<td>10053365</td>
<td>REV. 1</td>
<td>35A</td>
<td></td>
<td>AGAS II Honeywell CAS 67B Change 7.1: Rev. 1: Transfer from SAAB AB to Nychiping to SAAB AB.</td>
<td></td>
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<td>Active</td>
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<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>10053377</td>
<td></td>
<td></td>
<td></td>
<td>ATP Replacement Parts for Cargo Loading Overflow System in Single TI Aircrafts</td>
<td>See Certification Plan ATP01-CPL01 Edition 10 (CS-29 Cert Basis)</td>
<td></td>
<td>None</td>
<td>21/05/2015</td>
<td>Active</td>
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<tr>
<td>AVIONIK STRAUBLING ENTWICKLUNG</td>
<td>10053378</td>
<td>REV. 1</td>
<td>650</td>
<td></td>
<td>Installation of Aspen EFD1000H PFD System with ACU and RSM Panel-mounted EFS with Analog Converter Unit, Remote Sensor Module, and Configuration Module.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated anti-collision noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval</td>
<td>Flight Manual Supplement for Aspen Avionics EFD1000H Pro PFD System Doc No: ASR-2013-022-FMS-05-00. Prior to installation of the above listed documents approved by EASA. Project Description Doc. No: ASR-2013-022-PPS-04-04 for Continued Airworthiness Doc. No: ASR-2013-022-11-ICA-040/Instructions for Continued Airworthiness.</td>
<td>08/06/2015</td>
<td>Active</td>
<td>10003378</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>10053379</td>
<td></td>
<td>650</td>
<td></td>
<td>TCAS II Change 7.1 upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated anti-collision noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- List of Certification Documents. Scandanavian Avionics Design Agreeement No: 2091470-DDC, Issue 1, dated 28 April 2015. Aircraft Flight Manual Supplement. Scandanavian Avionics Design Agreeement No: 2091470-AM A, Issue 1, dated 28 April 2015. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/05/2015</td>
<td>Active</td>
<td>10003379</td>
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<tr>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>10053406</td>
<td></td>
<td>55, 55B, 55C</td>
<td></td>
<td>TCAS TPU 67A to B Upgrade to Version 7.1 (honeywell)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated anti-collision noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aerospace Flight Manual Supplement. AGH96/07, rev RInstructions for Continued Airworthiness - ICA, AG690-085. Project Description Doc. No: 2091470-DDC, Issue 1, dated 28 April 2015. Aircraft Flight Manual Supplement. Scandanavian Avionics Design Agreeement No: 2091470-AM A, Issue 1, dated 28 April 2015. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/05/2015</td>
<td>Active</td>
<td>10003406</td>
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<td>ASI AVIATION</td>
<td>10053414</td>
<td>REV. 1</td>
<td>B300</td>
<td></td>
<td>FAM 488 Initial Issue - KA350 SURMAR Mission equipment installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated anti-collision noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AMM Supplement MM-350-000-01 Rev 1 dated 13 May 2015POH Supplement (PO-350-995901 Rev 6 dated 13 May 2015STC Data List DL-350-000001 initial issue dated 20 May 2015or later revisions of the above listed documents approved by EASA. Limitation to serial numbers: FL-682, FL-786, FL-752, FL-777, FL-781, FL-800 and FL-802 equipped with radio kits no. 130-4015 HBC EASA STC1007658 and cabin audio system HBC FAA STCSA110600-C-D Before installation of the HBC FAA STC 5110600-C-D cabin audio system. If EASA validation shall be checked...</td>
<td>29/07/2015</td>
<td>Active</td>
<td>10003414</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>ROMTEX ANOUI AERONAUTIQUE S.R.L.</td>
<td>Cabin reconfiguration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Revision 0 dated 9 April 2015 of Airplane Flight Manual Supplement. Anpiusahoro document No. 02-40/1936-MDL-001, Revisions 0, 1 dated 9 April 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>28/05/2015</td>
<td>Active</td>
<td>10034345</td>
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<tr>
<td>NORTHERN AVIONICS S.R.L.</td>
<td>Installation of TV Communication Equipment for Shooting from Helicopter</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Rotorcraft Flight Manual Supplement RFM NAV505-02, Rev. 0 (applicable to helicopter versions AS355F1 &amp; AS355F2). Rotorcraft Flight Manual Supplement RFM NAV505-02, Rev. 0 (applicable to helicopter versions AS355NF) Instructions for Continued Airworthiness ICA-NAV505, Rev. 0 or later revisions of the above listed documents approved by EASA.</td>
<td>28/05/2015</td>
<td>Active</td>
<td>10053451</td>
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<td>FAA STC ST08954AC</td>
<td>Installation of tail floodlights on upper and lower surfaces of horizontal stabilizer</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Drawing Devore document No. 020000/05 Revision A, dated 16 January 1995 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of the above listed documents).</td>
<td>29/05/2015</td>
<td>Active</td>
<td>10053464</td>
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<td>AIRBUS S.A.S.</td>
<td>Airbus Corporate Jet Centre Modification CJ1880 VIP Project “Royal Thai Air Force”</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI-D-2322-001 - Minisuites CRI-E-21 - Isolated Compartments Equivalent Safety Finding(s): CRI-E-28 - Improved Flammability Standard Deviation (s): CRI-E-23...</td>
<td>- Modification Approval Sheet ref CJ1880, Issue 1, dated 28/05/2015 Airplane Flight Manual - AFM Supplement ref FM-S MOD CJ1880, dated 01/05/2015. Maintenance Planning Document - MPO Supplement ref MPO-S MOD CJ1800, dated 01/05/2015 or later revisions of the above listed documents approved by EASA.</td>
<td>01/06/2015</td>
<td>Active</td>
<td>10053471</td>
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<td>VOYAGEUR AVIATION CORP.</td>
<td>Fuel Quantity Gauging System Replacement - Auxiliary Fuel Tanks</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): E-01 – Fuel Tank Safety following paragraph(s) at a later amendment: CS 25.869, CS 25.869-CS 25.1401The requirements for environmental protection...</td>
<td>- Installation must be in accordance with TCCA approved Voyageur Airways Limited Design Change Summary DCS 13-1774, Rev. 0. Emissions for Continuing Airworthiness Maintenance must be in accordance with TCCA accepted Voyageur Airways Limited Maintenance Manual Supplement MMS 13-1774, Rev. 0.</td>
<td>01/06/2015</td>
<td>Active</td>
<td>10053477</td>
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<td>STC Holder</td>
<td>STC Holders</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>STC Ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>DART AEROSPACE Ltd.</td>
<td>SOCII - SH14-17</td>
<td>407</td>
<td>206L-1, 206L-3, 206L-4</td>
<td>206A, 206B, 206C</td>
<td>Replacement Engine Mounts.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Document List MDL-D407-811 Rev. A- Instructions for Continued Airworthiness (ICA)- D407-811 Rev. D. Installation Instructions INI-D407-811 Rev. B.</td>
<td>Installation of the Replacement Engine Mounts must be in accordance with Transport Canada Civil Aviation (TCCA) approved Dartz Aerospace Ltd installation instructions (IN- D407-811 Revision B) dated May 7, 2014.</td>
<td>01/06/2015</td>
<td>Active</td>
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<td>PMV ENGINEERING</td>
<td>B747-8</td>
<td></td>
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<td>Activation of eXPhone system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Document List SCA-00151-MDL Rev. 01 Certification Investigation report SCA-00151-CIR Rev 01 AFM Supplement SCA-00151-IFMS Rev 00 Instructions for continued airworthiness: MSN 37500-SCA-00151-ICA-00 Rev 00-MSN 37544-SCA-00151-ICA Rev 01</td>
<td>As per AFM supplement, the use of transmitting portable electronic devices (T-PEDs) is not allowed during any flight phase in the cockpit (( \Rightarrow )( \Rightarrow )) during critical flight phases (TTOL) in the cabin. Prior to installation of this design change it must be determined that the interrelationship between the...</td>
<td>02/06/2015</td>
<td>Active</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>A6365 N3</td>
<td>Solid State Voice Cockpit &amp; Flight Data Recorder Installation/MD MOD AEROTEC CONCEPT N°13114</td>
<td></td>
<td></td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CS29 amendment 3.29.25, 3.29.27, 2.25.21, 2.301, 2.303, 2.305, 2.25.61, 2.26.01, 2.26.05, 2.26.07, 2.26.09, 2.26.10, 2.26.25, 2.26.1301, 2.26.1309, 2.26.1301, 2.26.1303, 2.26.1307, 2.26.1309</td>
<td>Rotorkraft Flight Manual Supplement ref AE13114RFS, initial issue later revisions of the above listed documents approved by EASA MASTER DATA LIST document ref AE13114MDL Rev. 0, dated 2003/03/15 Valid for A330 N3 SN 6959 only Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not affect the environmental product.</td>
<td></td>
<td>02/06/2015</td>
<td>Active</td>
<td>10052010</td>
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<tr>
<td>AVIONICARE Ltd.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Upgrade the CAS 67A TCAS II Systems to Incorporate MOPS Change 7.1 functionality</td>
<td></td>
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<td>Applicable to aircraft equipped with a previously installed Bendix King CAS 67ATCAS II system.</td>
<td></td>
<td>02/06/2015</td>
<td>Active</td>
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<tr>
<td>Thomas Cook - Cabin rearrangement / IFE Installation/ Cabin seating rearrangement (&quot;Auto VIC&quot; new seats) and new Zodiac &quot;imsRAVE&quot; IFE system</td>
<td>A330-343</td>
<td></td>
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<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25 Amendment 15 - CS25.1131; CS 25.1301; CS 25.1303; CS 25.1305; CS 25.1307; CS 25.1309; CS 25.1310; CS 25.1311; CS 25.1312; CS 25.1313; CS 25.1314; CS 25.1315; CS 25.1316; CS 25.1317; CS 25.1318; CS 25.1319; CS 25.1320</td>
<td>Change Approval Sheet CAS-330-14-003 rev 0- Installation and Continued Airworthiness, ref: ICA-A330-14-003 rev 0</td>
<td>Prior to installation of this STC it is applicable to MSN 340, 336, 337 only. 2) The following modifications have to be embodied on the aircraft, prior to the requirements of the design of this STC: a) Lufthansa Technik Engineering Bulletin ref: A330-EB3-0391 revision0 for Guide U 1000 series photoluminescent emergency flare...</td>
<td>02/06/2015</td>
<td>Active</td>
<td>10052012</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>747-800 747-8</td>
<td>VIP Cabin interior completion/Revision 1: Replacement of conductive oxygen flex-hoses within non-conductive ones</td>
<td></td>
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<td>- The airplane shall be operated for private use only. - As per AFM supplement A50-1248- AFM section &quot;Certificate limitations&quot;. As per ICA document A-00-1248-ICA section 4 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other...</td>
<td>01/12/2020</td>
<td>Active</td>
<td>10052014</td>
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<tr>
<td>STC Holder</td>
<td>EMS &amp; Rescue Cabin Configurations in Active HELI-ONE System</td>
<td>SEE APPROVED MODEL LIST</td>
<td>STC Holder</td>
<td>EMS &amp; Rescue Cabin Configurations in Active HELI-ONE System</td>
<td>SEE APPROVED MODEL LIST</td>
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<td>CONCORDE BATTERY CORPORATION</td>
<td>C-212-DF, C-212-DE, C-212-CF, C-212-CD, C-212-DX, C-212-CB</td>
<td>Installation of Anti-Collision Light Systems</td>
<td>VOYAGEUR AVIATION CORP.</td>
<td>TCCA SA09-57 ISSUE 2</td>
<td>D-212-CB, D-212-DE, D-212-CF, D-212-CD, D-212-DX, D-212-CB</td>
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<td>Validation of FAA STC1787AT Replacement of Original Nickel-Cadmium Main and Auxiliary Batteries by Concorde Lead-Acid Batteries.</td>
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<td>Long Range Fuel System Retrofit</td>
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<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): E-01 – Fuel Tank Safety. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product</td>
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<td>AFM Supplement Nos. 5-0495, rev. NC, dated 30 September 2014 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/004/C (or subsequent revisions after this decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement Master D-</td>
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<td>Installation must be in accordance with TCCA approved Voyageur Airways Limited Design Change Summary DCS 09-0001, Rev. B. Flight operations must be conducted in accordance with TCCA approved Voyageur Airways Limited Flight Manual Supplement AFMS 09-0001, Rev. B. Instructions for Continuing Airworthiness...</td>
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<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>The installation of EASA STC 10053477 (TCCA STC SA14-58, Issue 2) is a prerequisite. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>Model</td>
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<td>1005354</td>
<td>ARMSTRONG AEROSPACE INC.</td>
<td>FAA STC</td>
<td>03</td>
<td>ST03398CH</td>
<td>A320-232</td>
<td>This STC installs a complete cabin interior consisting of:</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Armstrong Aerospace Master Data List, Doc. No. DC104-9501-00, Revision C, dated November 18th 2014, and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/06/2015</td>
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<td>10053544</td>
<td>LUFTHANSA TECHNICK AG</td>
<td>A320-232</td>
<td>02</td>
<td>BD-100-1A10</td>
<td>A321-232</td>
<td>This STC installs a complete cabin interior consisting of:</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Armstrong Aerospace Master Data List, Doc. No. DC104-9501-00, Revision C, dated November 18th 2014, and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/06/2015</td>
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<td>10053558</td>
<td>JET AVIATION AG</td>
<td>A340-642</td>
<td>05</td>
<td>BD-100-1A10</td>
<td>BD-100-1A10</td>
<td>The modification consists in the installation of folding step ladder/broom access to the tailcone equipment compartment in accordance with MPM Inc. All Ladder Installation Drawing LD001 dated 4th September 2003 (or later EASA approved revision).</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Master Data List, ref: JBSL-M-00003AFM-01 Issue 1, dated 06 June 2020, Aircraft Flight Manual Supplement, ref: JBSL-M-00003AFM-01 Issue 2, dated 25 March 2020, - Instructions for Continued Airworthiness, ref: JBSL-M-00003ICA Issue 1, dated 12 May 2015 - Weight and Balance Manual Supplement, ref: ALB...</td>
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<td>10053556</td>
<td>MACHINING PROGRAMMING MANUFACTURIN Gmbh</td>
<td>BD-100-1A10</td>
<td>03</td>
<td>CHALLENGER 300</td>
<td>BD-100-1A10</td>
<td>The modification consists in the installation of folding step ladder/broom access to the tailcone equipment compartment in accordance with MPM Inc. All Ladder Installation Drawing LD001 dated 4th September 2003 (or later EASA approved revision).</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Master Data List, ref: JBSL-M-00003AFM-01 Issue 1, dated 06 June 2020, Aircraft Flight Manual Supplement, ref: JBSL-M-00003AFM-01 Issue 2, dated 25 March 2020, - Instructions for Continued Airworthiness, ref: JBSL-M-00003ICA Issue 1, dated 12 May 2015 - Weight and Balance Manual Supplement, ref: ALB...</td>
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### List of EASA Supplemental Type Certificates

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<th>Rev. Level</th>
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<th>Model Numbers</th>
<th>Description</th>
<th>Certification Basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10053569</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 2</td>
<td>FAA SAO0907W-V-D</td>
<td>See EASA APPROVED MODEL LIST</td>
<td>Modification of a Rockwell Collins Inc. TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>The Certification Basis for the original product as amended by the FAA STC holder(s) under the FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>Installation in accordance with Rockwell Collins, Inc. FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>Approval of this change in type design applies to the aircraft in the attached Approved Model List (AML). Only A copy of this Certificate and the Approved Model List (AML) must be maintained as part of the permanent records for the corrected aircraft. Prior approved installation of a Rockwell Collins, Inc. FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>27/10/2015</td>
<td>Active</td>
<td>1053569</td>
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<tr>
<td>10053570</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 3</td>
<td>FAA SAO0907W-V-D</td>
<td>See EASA APPROVED MODEL LIST</td>
<td>Modification of a Rockwell Collins Inc. TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>The Certification Basis for the original product as amended by the FAA STC holder(s) under the FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>Installation in accordance with Rockwell Collins, Inc. FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>Approval of this change in type design applies to the aircraft in the attached Approved Model List (AML). Only A copy of this Certificate and the Approved Model List (AML) must be maintained as part of the permanent records for the corrected aircraft. Prior approved installation of a Rockwell Collins, Inc. FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>27/10/2015</td>
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<td>10053571</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>FAA SAO0907W-V-D</td>
<td>See EASA APPROVED MODEL LIST</td>
<td>Modification of a Rockwell Collins Inc. TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>The Certification Basis for the original product as amended by the FAA STC holder(s) under the FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>Installation in accordance with Rockwell Collins, Inc. FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>Approval of this change in type design applies to the aircraft in the attached Approved Model List (AML). Only A copy of this Certificate and the Approved Model List (AML) must be maintained as part of the permanent records for the corrected aircraft. Prior approved installation of a Rockwell Collins, Inc. FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>27/10/2015</td>
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<td>10053580</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>FAA SAO0907W-V-D</td>
<td>See EASA APPROVED MODEL LIST</td>
<td>Modification of a Rockwell Collins Inc. TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>The Certification Basis for the original product as amended by the FAA STC holder(s) under the FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>Installation in accordance with Rockwell Collins, Inc. FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>Approval of this change in type design applies to the aircraft in the attached Approved Model List (AML). Only A copy of this Certificate and the Approved Model List (AML) must be maintained as part of the permanent records for the corrected aircraft. Prior approved installation of a Rockwell Collins, Inc. FAA STC holder(s) for TCAS II System Installation to Change 7.0 or Change 7.1.</td>
<td>27/10/2015</td>
<td>Active</td>
<td>1053569</td>
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**Notes:**
- The STC holder is the primary entity responsible for the installation and maintenance of the STC.
- Rev. level refers to the revision level of the STC.
- Foreign/NAA STC ref. indicates any additional STCs or approvals from other authorities.
- Model Numbers include relevant model specifications.
- Description outlines key aspects of the STC installation.
- Certification basis details the regulatory requirements.
- Associated Technical Documentation, Limitations/Conditions, Date of Issue, Status, and System Record provide additional context for approval and implementation.

**EASA.europa.eu:**
- A website for European aviation regulations and certifications.
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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<th>Date of Issue</th>
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<td>10053586</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST00983L-A-D</td>
<td>737-400, 737-700, 737-800</td>
<td>Multi-Mode Receiver Installation</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements FAR 25 Amendment Original FAR 25.661, FAR 25.664, FAR 25.1301, FAR 25.1357, FAR 25.1641, FAR 25.1645, FAR 25 Amendment 25-22c10053604; FAR 25.301, FAR 25.303, FAR 25.611, FAR 25 Amendment 25-...</td>
<td>Rockwell Collins Aircraft Flight Manual Supplement 2508-AMI Revision A</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/06/2015</td>
<td>Active</td>
<td>10053580</td>
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<td>10053593</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>737-500 737-300</td>
<td>GPS Antenna Installation</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements CS 25.303, CS 25.307, CS 25.327, CS 25.317, CS 25.587, CS 25.601, CS 25.603, CS 25.605, CS 25.609, CS 25.1301, CS 25.1357, CS 25.1641, CS 25.1645, CS 25 Amendment 25-22c10053604;</td>
<td>Engineering Order &amp; Certification Program, Doc ref: 2094034-ED Issue 0, Compliance Record, Doc ref: 2094034-CC Issue 1 (continued Airworthiness, Doc 2094034-C4A issue 19 Structural Load Analysis Appendix, doc 2094034-5L, Issue 1 Structural Load Analysis Appendix, doc 2094034-5L, APP A issue 1 Technical Ord...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/06/2015</td>
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<td>10053604</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 1</td>
<td>A330-243</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements JAR 25.603 Change 14, JAR 25.613 Change 14, JAR 25.633a(s)(c) Change 13(14), JAR 25.634 Change 14, JAR 25.855a(s)(c)</td>
<td>Change Approval Sheet CAS-A330-14-005 rev 0 Instruction for Continued Airworthiness, ref ICA-A330-14-005-001 rev 0 (for later revisions of the above listed documents approved by EASA).</td>
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<td>01/10/2020</td>
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<td>10053613</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>EC 130 T2</td>
<td>E350-QCM34-001 Installation of an Avidyne TAS605 TAS removed - (New Installation) Avidyne TAS 605 Traffic Advisory System (TAS)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>Flight Manual Supplement: E350-QCM34-001-001-HMF-R2, dated 28.05.2015 for later revisions of the above listed documents approved by EASA Master Document List: E350-QCM34-001-R1, Section 2, dated 10.06.2015 Instructions for continued airworthiness: E350-QCM34-001-001-HMF-R2 dated 16.03.2015</td>
<td>Limited to S/N 7989 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>HEUPRODUCTS INDUSTRIES Ltd.</td>
<td>TCCA STC SH13-50</td>
<td>AS 355 E/F/F1/F3/F3NP AS 350 B/B1/B2/B3/B4/D Installation of AVEO Engineering position and antiluxionLED lights</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>Master Drawing List no. RLM-MDL-002 Rev A dated 05 August 2014 (AS 350series) Master Drawing List no. RLM-MDL-002 Rev A dated 04 November 2014 (AS 351series) Installation instructions no. HP350LEDXT-II REV 2 dated 10 January 2014 Installation instructions no. HP350LEDXT-II REV 3 dated 21 October 2014...</td>
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<td>Limitations/Conditions</td>
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<td>HELI-ONE</td>
<td>FAA STC</td>
<td>TCA STC SH12-26</td>
<td>AS 332 L1</td>
<td>Installation of Powersonix Loud Hailer</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement CHA332-25-045 FMS Rev. 3, dated 10 August 2012 or later revisions of the above listed documents/approved by EASA in accordance with the Technical Implementation Procedures of EU/Canda Bilateral Agreement Master Drawing List CHA332-25-045 MDL Rev. C, dated 23 October 2013 Comp.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a ground effect upon the airworthiness of the product.</td>
<td>11/06/2015</td>
<td>Active</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC</td>
<td>TSTO235AT-D</td>
<td>GULFSTREAM G280</td>
<td>Installation of a Heads Up Display (HUD) and Enhanced Vision System (EVS) on the Gulfstream Aerospace LP Model Gulfstream G280 airplane.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (a)/(C)/CR F-01 EYVS &amp; HUD for operational credit down to 100 ft. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Plan G250-GER-1054, Revision D, dated 27 January 2013; Aircraft Flight Manual Supplement G250-GER-1043, Revision B, dated 28 April 2014; Instructions for Continued Airworthiness G250-GER-0145, Revision B, dated 14 of May 2013; Instructions for Continued Airworthiness G250-GER-0027, Revision A, dated 23 October 2013</td>
<td>None</td>
<td>30/06/2015</td>
<td>Active</td>
<td>10053620</td>
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<td>GARMIN INTERNATIONAL, Inc.</td>
<td>FAA STC</td>
<td>SAE1000E</td>
<td>GARMIN GTN 6XX/7XX and GTN X4 Series Navigation System</td>
<td>Update of new hardware models for obstacle parts.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Flight Manual Supplement G250-GER-0145, Revision B, dated 28 April 2014; Instructions for Continued Airworthiness G250-GER-0145, Revision B, dated 14 of May 2013; Instructions for Continued Airworthiness G250-GER-0027, Revision A, dated 23 October 2013.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a ground effect upon the airworthiness of the product. The STC is valid for aircraft types and models as specified in the Master Flight Manual.</td>
<td>10/12/2021</td>
<td>Active</td>
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<td>10053659</td>
<td>PMV ENGINEERING</td>
<td>FAA STC</td>
<td>A310-322, A310-304</td>
<td>A310-325, A310-221, A310-222, A310-308, A310-203, A310-204, A310-324, A300-605R, A300 F4-622R, A300 F4-623, A300 B4-601, A300 B4-600, A300 622, A300 623, A300 605R, A300 605R, A300 F4-622R, A300 B4-601, A300 B4-600, A300 B4-622, A300 B4-623, A300 B4-605R, A300 B4-605R, A300 B4-605, A300 B4-622, A300 B4-623, A300 B4-605R, A300 B4-605R, A300 B4-605, A300 B4-622</td>
<td>Installation of Garmin GTN 6XX/7XX and GTN X4 Series Navigation System</td>
<td>Update of new hardware models for obstacle parts.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List AV1-00720-MDL-R01, dated 8 June 2015; Aircraft Flight Manual Supplement AV1-00720-AFSMS-R00, dated 4 May 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a ground effect upon the airworthiness of the product.</td>
<td>06/10/2015</td>
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<td>Foreign/NAA STC ref.</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>10053663</td>
<td>AERO-DEunist GMBH &amp; CO. KG</td>
<td>31, 31A</td>
<td>TCAS TPA71A to B Upgrade to Version 7.1 (Honeywell)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Equipment Supplement, Doc. No. AG991-071, Rev.</td>
<td>None</td>
<td>12/06/2015</td>
<td>Active</td>
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<tr>
<td>10053664</td>
<td>TATENHILL AVIATION Ltd.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>PA-31T2 (CHEYENNE IIO)</td>
<td>Installation of Various Aeronautical Antennas</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following CS-23 AMDt 3 paragraphs apply: 23.1, 23.23, 23.25, 23.29, 23.141, 23.171, 23.221, 23.301, 23.303, 23.305, 23.307, 23.321, 23.339, 23.337, 23.341, 23.351, 23.36.</td>
<td>Installation and Maintenance according to Tatenhill Aviation Ltd Installation Manual (document M00115 Revision A) or later revisions of the above listed documents approved by EASA.</td>
<td>- Limited to model PA-31T2 (Cheyenne IIO). To be installed in conjunction with the Tatenhill Minor Change TAL-00915. Installed antennas shall be maintained according to the Appendix A Section 3 of the Tatenhill Aviation Ltd Installation Manual (M. 00115 Revision A). Prior to installation of..</td>
<td>12/06/2015</td>
<td>Active</td>
<td>10553664</td>
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<tr>
<td>10053666</td>
<td>DIF STIFTUNG LUFRTRETTUNGS EMMENHUEZIGE AG</td>
<td>MBB-BK 117 C-1 MBB-BK 117 B-2</td>
<td>Installation of GNC 25S and GTN 650 COMM/NAV System without LPV/ SBA Capability</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>MBB-BK117 B-2/C-1 FMS Version 6.0.5 Initial Issue dated 05.06.2011 or later revisions of the above listed documents approved by EASA.</td>
<td>-GPS/RNAV approaches are prohibited Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>12/06/2015</td>
<td>Active</td>
<td>10553666</td>
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<tr>
<td>10053672</td>
<td>AVIONICARE Ltd.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BAE 146 SERIES 300 BAE 146 SERIES 200 BAE 146 SERIES 100 AVRO 146-RJ75 AVRO 146-RJ100 AVRO 146-RJ115</td>
<td>Upgrade of TCAS II Systems to incorporate MOPS Change 7.1 Functionality</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 12/25.1; CS 25.25; CS 25.27; CS 25.1301; CS 25.1309; CS 25.1622; CS 25.1431; CS 25.1601; CS 25.1519; CS 25.1529; CS 25.1581; CS 25.1585.</td>
<td>Avionicare Flight Manual Supplement FMS-0033-Ba4146(Avro146)RJ100/Revision A for later revisions of the above listed documents approved by EASA.</td>
<td>Applicable to aircraft with a previously installed TCAS II system incorporating an RMI-4850.TCR-150 or TCR-151 or a Rockwell Collins TTR-920 or TTR-921 or a Honeywell TPA-81A TCAS processor Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>15/06/2015</td>
<td>Active</td>
<td>10553672</td>
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<td>10053717</td>
<td>ALIDAUNIA S.R.L.</td>
<td>A109 All</td>
<td>Installation of swinging landing/search light</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>RFM Supplement FMS/ALU/ 024, Rev. 0 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>17/06/2015</td>
<td>Active</td>
<td>10553717</td>
<td></td>
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<tr>
<td>10053718</td>
<td>LAZ-TEC, LLC</td>
<td>SR00780ED</td>
<td>Installation of Tail Rotor Pedal Removal Kit/This STC installs a kit to allow the rapid removal of the Tail Rotor Pedals at the left hand pilot position when a non-pilot occupies the seat.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Drawing list – Report Number LT-13507 Revision 1 Installation Instructions – Report Number LT-13507 Revision 1 Instalation for Continued Airworthiness – Report LT-13502 1. FAA accepted Instructions for Continued Airworthiness, Laz-Tec, LLC Document No. LT-13202, No Revision, dated May 30, 2014, FAA accepted June 02, 2014, or later FAA accepted revision is required.2. If the STC holder agrees to permit another person to use this certificate to alter the product, the holder..</td>
<td>17/06/2015</td>
<td>Active</td>
<td>10553718</td>
<td></td>
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</table>
10053719 | AERODATA AG | Models 737-700, 737-800 | Installation of Aerodata "AeroPhone+" Satcom communication system. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. FAA STC 10053726 | SPECTRUM AEROMED, LLC/ b/a SPECTRUM AEROMED | REV. 2 FAA ST01661CH | LEARJET 45 | Air Ambulance Conversion | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Prior to installation of this design change it must be determined that this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. FAA STC | Active | 10053719 | 17/06/2015 | 10053726 | 06/09/2021 | REV. 1 FAA ST01138CH | 560XL | Air Ambulance Conversion | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. - Prior to installation drawing 6015, Revision 4 dated October 15, 2012: Per Engineering change description SII MM-1321 7/11. FAA STC | Active | 10053732 | 23/04/2019 | 10053763 | 22/06/2015 | FAA STC | SAD244F1K | 7600 | Installation in accordance with: Master Data List 6050-XLMM Rev. 5 dated November 29, 2012: Aircraft Flight Manual Supplement FMS 6015, Revision 4 dated August 14, 2014: Operation in accordance with: Flight Manual Supplement, PA32- GA8LA, Rev(1), dated August 14, 2014: Instructions for Continued Airworthiness for Continued Airworthiness including installation, modification and repair of this product. AERGLAS, Inc. | Active | 10053763 | 22/06/2015 | DISCOVERY AIR TECHNICAL SERVICES | G58 | Installation – Intelligence, Surveillance and Reconnaissance (ISR) Special Mission System | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. FAA STC | Active | 10053769 | 22/06/2015 | G58 | Installation – Intelligence, Surveillance and Reconnaissance (ISR) Special Mission System | The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. FAA STC | Active | 10053769 | 22/06/2015
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<th>STC Holder</th>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FA A1</td>
<td>777-100UR</td>
<td>Panasonic System 3000 in-Flight Entertainment System</td>
<td>DELTA ENGINEERING MASTER DATA LIST NO. 1043-06/76-2215 Revision J Jattered2 January 2013</td>
<td>Approved configurations are: TL 10593-01+, &lt;TL 10593-02, TL 10593-03, and TL10593-04. This STC includes activation of passenger seat Panasonic in-flight entertainment system (IFE) equipment. The conditions of activating these IFE as follows: The seat IFE equipment as defined on Panasonic drawing ISL.</td>
<td>23/02/2015</td>
<td>Active</td>
<td>10053781</td>
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<tr>
<td>STIRLING DYNAMICS GMBH</td>
<td>REV. 4</td>
<td>737-400</td>
<td>Complete seat replacement to install Zodiac Dragonfly 31 Series Seats.Rev: 4 Transfer of STC</td>
<td>214210-C-CL-2520-001-02Master Document List 214210-C-CL-2520-001-02 later revisions of the above listed documents approved by EASA/System Description Doc. 214210-C-DE-2520-001-01Drawing Register Doc. 214210-C-DR-2520-001-01</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2022</td>
<td>Active</td>
<td>10053782</td>
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<tr>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SR</td>
<td>412EP</td>
<td>Left Hand Alt Equipment Shelf Kit P/N 412-340-001Installation of a LH Equipment Shelf, in the left hand aft cargo compartment on Bell Helicopter Textron 412 EP helicopters.</td>
<td>The Certification Basis (CB) for the original product remains applicable: this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation Drawing List Document AA-10019 Rev. NR, Installation Instructions Document AA-10020 Ref. NR, Instructions for Continued Airworthiness Document AA-10018 Rev. NR,</td>
<td>23/06/2015</td>
<td>Active</td>
<td>10053787</td>
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<tr>
<td>BERINGER AERO REV. 4</td>
<td>DA 42</td>
<td>Installation of FF-006(A) main wheel and RA-014(A) nose wheel with ISA-003.7NA(A) brake system on Diamond Aircraft DA-42 models</td>
<td>Installation of FF-006(A) main wheel and RA-014(A) nose wheel with ISA-003.7NA(A) brake system on Diamond Aircraft DA-42 models.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Modification file: DM-MOD001-STC-013 Rev 00 dated 12 Oct 2023, Standard instructions for Continued Airworthiness and Installation Instructions. BERINGER wheels and brakeson DA 42 aircraft: ICA-ETC/013 Rev 04 dated 09 Aug. 2003, later revisions of the above listed document(s) approved/accepted under the EAS.</td>
<td>13/10/2023</td>
<td>Active</td>
<td>10053795</td>
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For more information, visit [easa.europa.eu](http://easa.europa.eu).
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<tr>
<td>ALPINE AEROTECH LP</td>
<td>A321-212/-213/-231/-232</td>
<td>Active FAA STC 10053796 for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Alpine Aerospace LLC, Master Document List 191M-96-028-MDL, Rev.NI</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/06/2017</td>
<td>Active</td>
<td>10053796</td>
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<tr>
<td>AVIONITEC AG</td>
<td>M20J</td>
<td>The Certification Basis for the original product as amended to include the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment CS-23 Initial Issue 23.301, 581(1), 1301, 1309, 1321, 1325, 1327, 1329, 1351(1), 1352, 1355, and 1356(1).</td>
<td>APM Supplement Power Flarm No. 5322-011 of 25.04.2014 or later revisions of the above listed documents as approved by EASA Master Document List No. 5322-017 Rev. 0 of 03.04.2015 for Continued Airworthiness No. 5322-015 of 20.04.2015</td>
<td>Limited to S/N 24-0775 HB-DFM</td>
<td>24/06/2015</td>
<td>Active</td>
<td>10053790</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN AS</td>
<td>S550 (CITATION VII)</td>
<td>The Certification Basis for the original product as amended to include the following additional or alternative airworthiness requirements: CS 25 amdt 4CS 25.1, CS 25.1301, CS 25.1309, CS 25.1501, CS 25.1529, CS 25.1581. The requirements for environmental protection and the associated certificated noise and/ or emissions.</td>
<td>Engineering Order &amp; Certification Program - Doc ref: 2053368-EO, Issue dated 11 June 2015</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2015</td>
<td>Active</td>
<td>10053815</td>
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<tr>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td>A321-212/-213/-214/-215</td>
<td>Installation of Dual Avionics Class 3 Electronic FlightBagRevision 1 is a correction of the applicable models</td>
<td>Master Data List, Aviation Support Group document No. 29106-MDL, Revision 1(c) dated 24 October 2014 for Continued Airworthiness, Aviation Support Group</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/01/2021</td>
<td>Active</td>
<td>10053816</td>
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<tr>
<td>LEARJET Inc.</td>
<td>FAA STC 10053827</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC 10053827-01. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List BAS45010024, Revision &quot;B&quot;, dated 3 March 2014; FAA approved on 3 March 2014, or later FAA approved revision Airplane Flight Manual Supplement BAS45040108 Revision NC. FAA approved 24 March 2014, or later FAA approved Revision (required for the -1 &amp; -2 Configurations Airplane Flight Manual.</td>
<td>For the -1 Configuration: A prior or concurrent requirement modification is the installation of an approved Dual Universal Avionics Systems Corporation (UASC) UMS-108 (WiAAS) FMS with S/N 100-05 and approved Enhanced Surveillance System for aircraft without HF capability. For the -2 Configuration...</td>
<td>29/07/2021</td>
<td>Active</td>
<td>10053827</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>REV. 5</td>
<td>FAA ST0269NY</td>
<td>MYSTÈRE FALCON 900 (MF900) FALCON 900EX (P900EX)</td>
<td>Validation of FAA STC ST0269NY installation of triple Honeywell International, Inc. Flight Management System (FMS) v7.1 Upgrade Revision 1: Correction of pre-requisite FAA STC number Revision 2: Correction of STC Holder name Revision 3: Correction of STC Holder address: Revision 4: Installation of ...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC ST0269NY. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Data List: FAA F9XW-D0004-005, Revision A, dated 3 September 2013, or later FAA approved revision. - Instructions for Continued Airworthiness (ICA) FAA F9XW-D0004-155, Revision B, dated 2 July 2013, or later FAA approved revision. - Airplane Flight Manual Supplement F9XW-D0004-150, Revision IR, dated...</td>
<td>The installation of the EASA STC 10051492 (validation of FAA STC ST0269NY) for the installation of Honeywell GPS Landing System Sensor Unit (GLSSU) is a pre-requisite for this design change. Prior to installation of this design change it must be determined that the interrelationship between this design...</td>
<td>26/02/2020</td>
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<tr>
<td>ASI INNOVATION</td>
<td></td>
<td></td>
<td>B200</td>
<td>Retractable mast with relay antenna installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with ASI Innovation Master Drawing List ref:ASI-MDL-0299 Rev 2, dated 02/06/2015 and Service Bulletin ASI-SB-B200-2300-0299 Revision 1, dated 01/06/2015; Operation in accordance with ASI Innovation Flight Manual Supplement ref:ASI- AFM-0299 Rev 0, dated 11/05/2015; Instruction...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/06/2015</td>
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<tr>
<td>CESSNA DUSSELDORF CATION SERVIC E CENTER GmbH</td>
<td></td>
<td></td>
<td>650 (CITATION III, VI, VII)</td>
<td>Certification of P-RNAV capability of aircraft equipped with Flight Management System Honeywell FMZ-2000</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Document List, Cessna Düsseldorf document No. 2014-15-16, Revision IR, dated 02 December 2014 - Airplane Flight Manual Supplement, Cessna Düsseldorf document No. 2014-15-8, Revision IR, dated 02 December 2014 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/06/2015</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>REV. 1</td>
<td>FAA ST0269NY</td>
<td>MYSTÈRE FALCON 900 (MF900) FALCON 900EX (P900EX)</td>
<td>Honeywell FMS CD-830 CTL Display Unit (Revision 1: Removal of existing Honeywell CD-810 or CD-820 Control Display Units (CDUs) and replaced with Honeywell CD-830 CDUs.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any...</td>
<td>Master Data List FAA F9XW-D0002-005 Rev C, March 21, 2017; Instructions for Continued Airworthiness (ICA) FAA F9XW D0002-155, Rev B, March 15, 2017; or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>None</td>
<td>18/05/2020</td>
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<tr>
<td>EIOTECH AVIATION SERVICES LIMITED</td>
<td>REV. 1</td>
<td></td>
<td>737-800</td>
<td>Approval for PED Operation on Boeing 737 Aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product as unchanged and remain applicable to this certificate/ approval.</td>
<td>None for later revisions of the above listed documents approved by EASA</td>
<td>Limited to B737-800 aircraft MSHs listed in FMS/ 688-015 Prior to installation of this modification it must be determined that the interrelationship between this modification and another previously installed modification and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>09/04/2015</td>
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<td>EASA STC No.</td>
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<td>10053848</td>
<td>BOLDER MANUFACTURING G.L.C.</td>
<td></td>
<td>FAA STC ST04717CH</td>
<td>767-300F 767-300</td>
<td>Installation of Galley A4 and F3</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List MDL-2 BM141001M rev. NC</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10053856</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td></td>
<td>FAA STC SR00254AT</td>
<td>212412/412EP</td>
<td>Quick Mount Kit for the Spectrova Nightsh SX-16 Searchlight</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement AA-69207 revision D or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions offer this decision).</td>
<td>29/06/2015</td>
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<td>10053859</td>
<td>ROEDER PRAEZISION GMBH</td>
<td>REV. 1</td>
<td>A320-214, A320-216</td>
<td></td>
<td>Activation of PED power system on Recaro 3500 seats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Report A320-CRS5-03-2015 Rev. 01, dated 28/06/2015 Master Documentation List A320-MDL25-03-2015 Rev. 00, dated 28/06/2015 Instructions for Continued Airworthiness A320-ICA25-03-2015 Rev. 01, dated 28/06/2015 Aircraft Flight Manual Supplement A320-AF525-03-2015 Rev. 00, dated 28/06/2015 or...</td>
<td>None</td>
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<td>10053886</td>
<td>AIR ALLIANCE GmbH</td>
<td></td>
<td>35, 35A</td>
<td></td>
<td>Installation of Instrum SATCOM Antenna to support SituationAwarenessTools</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Air Alliance, Instructions for Continued Airworthiness 2015_001_L35, 0507, rev 00, dated 24 June 2015 or later revisions of the above listed documents approved by EASA</td>
<td>None</td>
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<td>10053887</td>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td></td>
<td>FAA STC ST04916AT</td>
<td>767-300</td>
<td>Installation of an Interior Reconfiguration in a Boeing 767-300</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>HAECO Americas Cabin Solutions Master Data List 1470681001 Revision B dated 28 June 2015</td>
<td>None</td>
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<td>10053890</td>
<td>AERODATA AG</td>
<td></td>
<td>CL-600-2B16</td>
<td></td>
<td>FMS Upgrade UNS-1B</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CS-25 Amendment 1525.60; 25.771; 25.899; 25.1301-25.1302; 25.1309; 25.1329; 25.1336; 25.1431; 25.1501; 25.1526; 25.1581; 25.1583; The requirements for environmental protection...</td>
<td>Aerodata Airplane Flight Manual Supplement AFM-149 Rev for later revisions of the above listed documents approved by EASA</td>
<td>None</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>10503893</td>
<td>BOGGI S.R.L.</td>
<td></td>
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<td>AW139</td>
<td>Installation of incubator ATOM V808 on AW139 helicopter - change 139P010-00</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CS 29 Amdt. 3 for changed parts. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Flight Manual Supplement - 139P0100-9RM6-01, Issue 0.</td>
<td>–</td>
<td>01/07/2015</td>
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<td>10503896</td>
<td>BOGGI AERONAUTICS S.R.L.</td>
<td>REV. 2</td>
<td></td>
<td>MBB-BK117 D-3 MBB-BK 117 C-2, MBB-BK 117 D-2</td>
<td>Installation of incubator ATOM V808 on MBB-BK117 D-2, change 14P001-00. Variants MBB-BK117 C-2s and MBB-BK117 D-2m are not eligible for this STC. Rev. 1 of this STC is including a new structural interface for incubator. PIN 14P001-10 and expand the type of cabin door config.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraphs at a later amendment: CS 29 Amdt. 3 for the affected areas. For the change 14P001-30 (a. Rev. 2 of this STC), CS 29. Amdt. 4 for the following re...</td>
<td>Initial issue for the change 14P001-00: Flight Manual Supplement 14P00100-RFMS-02, Issue 2 Instruction for Continued Airworthiness 14P00100-ICA-00, Issue 0 Rev. 1 for the change 14P001-01: Flight Manual Supplement 14P00101-RFMS-01, Issue 1 Instruction for Continued Airworthiness 14P00101-ICA-00...</td>
<td>14/10/2022</td>
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<td>10503897</td>
<td>OGMA INDUSTRIA AERONAUTICA DEPORTUGAL, S.A.</td>
<td>REV. 1</td>
<td></td>
<td>EMB-145M, EMB-151R EMB-145EU, EMB-145I, EMB-145P, EMB-151ER, EMB-141LR, EMB-135BJ, EMB-141ER</td>
<td>Installation of FBG mounting device. The installation as per this STC can result in two configurations: A. Installation on the not assembly of the cockpit LH or RH direct-vision windows. B. Installation on the frame of the cockpit LH or RH direct-visionwindows.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AFM Supplement (to Embraer AFM 145/154, AFM 135/123 and AFM 135/143).</td>
<td>Prior to installation of this design change it must be determined that there is no interrelationship between this design change and any other previously installed design change and/or repair will introduce a adverse effect upon the airworthiness of the product.</td>
<td>02/07/2015</td>
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<td>10503902</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td></td>
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<td>A300F4-608ST</td>
<td>Installation of TCAS 7.1 on A300-600ST fleet. The modification consists in: Removal of current Collins TCAS TTR-800 standard 7.0 (P) N622-8971-322&lt;(&gt;,&lt;)&gt;-,, Installation of new Collins TCAS TTR-821 standard 7.1 (P)N622-830-334.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airbus Transport International, Fiche de Modification, ST34MOD041, Issued 01, dated 01 June 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>02/07/2015</td>
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<td>10503904</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td></td>
<td>A380-841, A380-842, A380-861</td>
<td>LHT – A380 Installation of a Patient Transport Compartment.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-01 Installation of a Patient Transport Compartment (PTC): CRI H-01 – ICA on EWIS the requirements for environmental protection and the associated...</td>
<td>Supplemental Type CefDoc. (STDD025-1345M Iss.02 Rev.00) Compliance Record List (CRL)25-1345M - Iss.02 Rev.00 (Instructions for cont. Airworthiness (A-00-15-24-ICA Rev.1)) or ramifications of the above listed documents approved by EASA Applicants Decl. of Compliance (AdJC3566 - updated with Rev.02)...</td>
<td>PPW STC 10051116 is a prerequisite to install this STC. Prior to installation of this design change it must be determined that there is no interrelationship between this design change and any other previously installed design change and/or repair will introduce a adverse effect upon the airworthiness of the...</td>
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<td>STC ref.</td>
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<td>Description for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
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<td>AVM SOLUTIONS</td>
<td>10053922</td>
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<td>Dual Garmin GNS 430 Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>TCCA STC SA15-28 CL-215-EB1 (CL-415 VARIANT)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25.116c. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>365 AEROSPACE DDS LIMITED</td>
<td>10053925</td>
<td>REV. 1</td>
<td>CL-600-2B16</td>
<td>CL-600-1A1</td>
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<td>SATCOM Antenna Installation</td>
<td>The Certification Basis (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25.116c. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>AVIATION TRADERS LIMITED</td>
<td>10053931</td>
<td>REV. 2</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td></td>
<td></td>
<td>Introduction of the Smartcart Galley Cartfife 2: Transfer of STC</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: CS 25.116c. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>DLR DEUTSCHES ZENTRUM FUERUFT</td>
<td>10053932</td>
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<td>A320-232</td>
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<td>Operator seat double</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>AEROSPACE INTERIOR REFURBISHMENT ESPANA S.L.</td>
<td>10053948</td>
<td>REV. 2</td>
<td>A321-213/-211/-212</td>
<td>A321-111/-121/-122/-211/-212</td>
<td></td>
<td>Passenger Seating Replacement and In-Seat Power Supply System REV. 1 adds some models of the Airbus A319 series and A321 series aircraft to EASA STC 10053948 Rev. 2: certificate transfer from Tamencia Ltd. to Aire S.L.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10/11/2021</td>
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List of EASA Supplemental Type Certificates

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**10053651**

**FEDERATION FRANÇAISE DE VOL A VOILE (FFVV)**

**SMC 34C**

Method to install aeron hinges on Alliance 34 SN 34 C.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Sbulved by Service FFVV 01/2015 rev1 du 30 Jun 2015 or later revisions approved by EASA.

None

06/07/2015 Active 10053651

**10053652**

**HELI AUSTRIA GmbH**

**REV. 2**

412, AB412 412

HAU-D-010, Tarern an Hel Pilot relocation..This change moves the Tarern and Hall Pilot System including Terminal Blocks, Trim Computer into the nose compartment to reduce the required weights in the nose. No new equipment or functions implemented.Rev. 1 extends to all serial numbers of Model 412 a... The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.


None

16/11/2021 Active 10053652

**10053656**

**STANDARDO AERO**

**FAA STO2911CH-D**

G-IV

Installation of an Interior Reconfiguration.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Master Data List 1019483 rev B dated Feb. 01, 2011 AAM Supplement 101983 rev A dated Feb. 04, 2011 for later revisions of the above listed documents approved by EASA in accordance with EASA EO Decision 2004/454/CO (or subsequent revisions off this decision) and/or the Technical Implementation Procedures o...

None

07/07/2015 Active 10053656

**10053659**

**DELTA ENGINEERING CORPORATION**

**FAA STC ST02983NY**


Panasonic In-Flight Entertainment EN2 system installation on A330-200. This modification installs Panasonic x2-IFE providing audio, video and in-seat power supply system to the passengers for use in flight.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 at various amendments CS-25 (Amendment 11): 25.853(a)(b). The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.


Depending on the various configurations supported by this STC, the following modifications are prerequisites of this Certification TL1060-01 requires Luftansa Technica STC, ref. STDD001-20/SM (EASA approval 100489220-Configuration TL1060-02) requires Airbus Modifications 95628, and208687 (approved see...

07/07/2015 Active 10053659

**10053970**

**SOMMER, MICHAEL**

**PA-28-180 (ARCHER)**

Installation of DME KN82A and PM6000 for IFR operation.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23 Amendment 3, 23.21, 23.776(a), 23.1301, 23.1309, 23.1321, 23.1351 and/or 23.1356. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Installation in accordance with Change certification document for PiperCh-185. Ref. OWN-360-Decl-P.1 dated 20/05/2015 Operation in accordance with Aircraft Flight Manual Supplement 12.3 STC-OWN- DECL-P.1 dated 07/07/2015 or later revisions of the above listed documents approved by EASA.

Limited to serial number: S/N 28-4404

07/07/2015 Active 10053970

**10053383**

**Q.C.M. DESIGN GmbH**

**CL-600-2B16 (604 VARIANT)**

Change in Layout of Passenger Accommodation by relocation of a divan and entertainment rack.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.


None

08/07/2015 Active 10053383

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<th>EASA STC No.</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<td>10053993</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td>A01/0065P</td>
<td></td>
<td>A319-115</td>
<td>Installation of Entertainment System CD/ DVD Player with dual 7” foldable display, Drawing 23-NAV0493-00 SH 1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement RM5 NAS/0493, Rev. 3 or later revisions approved by EASA. Instructions for Continued Airworthiness and/or the certification basis are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/07/2015</td>
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<td>10053998</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>REV. 1</td>
<td></td>
<td>A330-300</td>
<td>Economy class seat installation for Cathay Pacific A330-300 Aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Summary, ref : N0495-D09 rev - Master Drawing List, ref - N0495-D10 rev Construction for Continued Airworthiness, ref : N0495-D10 rev A or later revisions of the above listed documents approved by EASA.</td>
<td>1) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/07/2015</td>
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<td>10054000</td>
<td>AIRBUS S.A.S.</td>
<td>A319-115</td>
<td></td>
<td>A330-300</td>
<td>Installation of a supplemental NVIS lighting system for Cathay Pacific Aircraft.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRRA-4301-019 Issue 1 “Certification basis for A330-300 MSN 5792” The requirements for environmental protection and the associated certified noise and/or emissions levels…</td>
<td>Modification Approval Sheet CJ2000, issue 1, dated 05/07/2015 Airplane Fight Manual – AFM Supplement FM-S MOD C-1960, dated 01/07/2015 Maintenance Planning Document – MOD Supplement/MOD-S MOD CJ1900, dated 01/07/2015 or later revisions of the above listed documents approved by EASA.</td>
<td>This approval is limited to MSN 5792.</td>
<td>08/07/2015</td>
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<td>10054006</td>
<td>AVIATION SPECIALTIES UNLIMITED, INC</td>
<td>FAA SR01385SE</td>
<td></td>
<td>206L-1, 206L-3, 206L-4, 206L-6</td>
<td>Installation of a supplemental NVIS lighting system Validation of FAA STC SR01385SE</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Aviation Specialties Unlimited, Inc. Master Drawing List (MDL) MDL2060044, Revision C, dated August 8, 2011 - Rotorcraft Flight Manual Supplement No. FMS-206004-6 for Bell Helicopter Textron Model 206 Series rotorcraft.</td>
<td>- Compliance with EASA, AD US-2012-12-11, 1 is mandatory - It is recommended that first installation is made with support from STC holder - EASA must be notified at time of first installation for check of compliances of applicable configuration EASA CRI F-01 on NVIS dated 25Feb 2015 selectors - Approval of...</td>
<td>29/07/2015</td>
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<td>10054019</td>
<td>DOA21 LIMITED</td>
<td>A320-214</td>
<td></td>
<td>A320-214</td>
<td>Introduction of VIP Interior Rev. 2: Transfer of STC holder role to DOA21 Ltd.</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (a) D-02 Isolated CompartmentD-05 Glass Screens of Displays/MonitorsD-06 Show/In/Installation/Deviation (a) D-01 Installation of Interior Doors.</td>
<td>STC Twenty One Aeroplane Flight Manual Supplement S21 FMS-0120 Issue 2STC Twenty One Supplement Manual S21.TEC-064 Issue 1, Airworthiness/Limitations (Appendix A to Comlux Maintenance Planning Document Supplement PRM-0000RM00 Revision 1) or later revisions of the above listed documents approved by EASA.</td>
<td>Any changes to the WLAN or interior layout approved under this design change will require re-examination to determine if compliance/or completion of applicable configuration EASA CRI F-01 on NVIS dated 25Feb 2015 rela...</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>New Premium Economy Class (B747-400) the main cabin of the main floor on the aircraft, the change consists of the removal of 10 per cent of the Economy Class Seats between door 2and door 3 and the installation of a New Developed Premium Economy Seatinstead. The new seating will have a greater pitch a...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>- Suplemental Type Definition Document STDO25-13/44M level 1, ICA ref A-00-10-36-ICA, Issue 2 rev 0, Compliance Record List CRL1S-13/44M K1 rev 0 or later revisions of the above listed documents approved by EASA.</td>
<td>According to the design related to this STC. This modification must be installed in conjunction with a) SB AIM B747-25-173 - Introduction of a three part class dividers installation/Minor Change ref MM 636-14(b) STC 15014325 (rev.7 or later revisions), Non Electric Floor PathMarking System, together wit...</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>New Premium Economy Class (A330-300) the main cabin of the main floor on the aircraft, the change consists of the removal of 10 per cent of the Economy Class Seats between door 2and door 3 and the installation of a New Developed Premium Economy Seatinstead. The new seating will have a greater pitch a...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>- Suplemental Type Definition Document STDO25-13/44M level 1, ICA ref A-00-10-36-ICA, Issue 2 rev 0, Compliance Record List CRL1S-13/44M K1 rev 0 or later revisions of the above listed documents approved by EASA.</td>
<td>According to the design related to this STC. This modification must be installed in conjunction with a) SB AIM A320-25-050 - Introduction of a three part class dividers installation/Minor Change ref MM337-14(b) STC 15014507 rev2 (or later revision). Non Electric Floor PathMarking System, together wit...</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>Installation of Forward Mount for the (FSI) FLIR Systems, Inc. S/4FIRE Imaging System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement AA-84/161 No Rev or later revision of the above listed document approved by EASA in accordance with EASA EDD/2004/04/CF (or subsequent revisions of this decision).</td>
<td>Bell Model 212 rotorcraft is limited to day or night VFR operations. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and or repair will introduce roadawe affect upon the aircraf...</td>
<td>15/07/2015</td>
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<td>ROMETEX ANJOU AERONAUTIQUE S.R.L.</td>
<td>B737NG Cabin Layout ModificationThe old seats have been removed; new BRICE seats have been installedwith increasing or decreasing passenger capacity. New class dividersinstallation some MSNs. On the remaining MSNs, the existing class/ dividers have been relocated. PSU channel layouts have been modif...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>DOA Technical Report 02-100003-DOTR-001 revision 0content equal to Certification Programme/PlanMaster Document List/02-100003-MDL-001 revision 0 or later revisions of this above listed document approved by EASA in accordance with EASA EDD/2004/04/CF (or subsequent revisions of this decision).</td>
<td>STC limited to MSN 28982, 28984, 28985, 28986, 28988, 30362, 28980,28983, 28987, 30365, 28981, 30368, 30369, 30360, 30361, 30363, 30364, 30365, 30366, 30367, 33071, 33068, 33069, 33070, 37718, 33073, 33074, 33075, 33079, 33622, 38626, 38629, 38628, 38631, 33976, 33978, 33979. Prior to installation...</td>
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<td>AVIONITEC AG</td>
<td>Avionics Upgrade1. List of removed units Manufacturer:.. Unit Function:.. Unit Name:Bendix/ King.. RG:.. NAV Indicator:.. R.. 8-133-006Aeritalia.. Vertical Speed md:.. R..,.. 8-041-506Aeritalia.. Attitude R.. NS/600-2Aeritalia..</td>
<td>The Certification Basis for the original product as amendedby the following additional or alternative avionics and related requirements: the paragraph(s) at a later amendment: CS 23.001: 23.001, 23.015, 23.121, 23.125, 23.127,(&gt;), 23.131, 23.135, 23.139, 23.141, 23.152, 23.159, 23.161..</td>
<td>AFMS for dual Garmin GNS 833/430 system Doc. No. 1362-011 Rev AFMS for Sandel EHSI 84000 Doc. No. 2536-012 Rev 15 12.2014AFMS for Sandel EASI 84000 Doc. No. 2536-013 Rev. 15 12.2014AFMS for Avionics TAE 615 Doc. No. 2536-014 Rev 15 12.2014 Rev of later revisions of the above listed documents approved...</td>
<td>Limited to S/N 2-58, HB- KOY Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and or repair will introduce roadawe affect upon the aworthiness of the product...</td>
<td>23/07/2015</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>10054089</td>
<td>S. E. I. INTERNATIONAL SERVICE ELECTRONIQUE INFORMATIQUE</td>
<td>REV. 2</td>
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<td>S-76C</td>
<td>Installation of Flight Data Monitoring and Transmitting System HELICOM2 NUM or HELICOM 2+ NUM Revision 1 to STC Correlation of Engine model/Revision 2 to STC - Proposed additional installations for the main device. GPH SATI GSM/interference and the Air/Ground sensor, Extension of applicability of system for...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): SC F-01. Lithium Battery Installations. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product have been amended and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement N11-576C+ HV2 HV2+ Num.Revision 04 or later revisions of the above listed document/approval by EASA. STC Twenty One Master Document List No. S21.MOL-126 ISSUE 2</td>
<td>For helicopters with engines Arrisol 21S and with or without IDS RDAU equipment installed. Prior to installation of this design change, it must be determined that the installation relationship between this design change and any other previously installed design change(s) or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>08/12/2015</td>
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<td>10054105</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 2</td>
<td></td>
<td>A319-115</td>
<td>Activation of WAP and Media Servers</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product have been amended and remain applicable to this certificate/approval.</td>
<td>STC Twenty One Aircraft Flight Manual Supplement S21 FMS-0118 Issue 3 or later revisions of the above listed documents approved by EASA STC Twenty One Master Document List No. S21.MOL-126 ISSUE 2</td>
<td>Prior to installation of this design change, it must be determined that the installation relationship between this design change and any other previously installed design change(s) or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>17/07/2015</td>
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<td>10054112</td>
<td>SCANDINAVIAN AVIONICS DESIGN A/S</td>
<td>REV. 1</td>
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<td>Installation of a Five Panel Universal EFIS-88R EFIS to BAE Systems AT5P Systems Number 2033 Major Change relates to extension of Approval Time Limit.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI F-01. HRF/CRI H-01 Enhanced Airworthiness Programme for Aeronautical Systems (EAPAS)/PAGE Instructions for Continued Airworthiness (ICA) on Electric...</td>
<td>Scandnavian Avionics Design Document 2093719-AM A, Approved Aircraft Flight Manual Supplement Initial, Issue 4 dated 22 June 2016; Scandnavian Avionics Design Document 2093719-AMC, Approved Aircraft Flight Manual Supplement, Issue 3 dated 02 June 2016 or later revisions of the above listed documents.</td>
<td>Prior to installation of this design change, it must be determined that the installation relationship between this design change and any other previously installed design change(s) or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>06/07/2016</td>
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<td>10054139</td>
<td>HELIFAS Inc dba ARRION AVIATION CO. LLC</td>
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<td>S-76A, S-76B, S-76C</td>
<td>Installation of a 62 Gallon Auxiliary Fuel Tank in the baggage compartment in a Sikorsky S-76A, B &amp; C Rotorcraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product have been amended and remain applicable to this certificate/approval.</td>
<td>- Rotorcraft Flight Manual Supplement, document N HFS-576-580-004 Revision 0 dated February 06, 2014 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decisions 2004/040/CF (or subsequent revisions of this decision) and/or the Technical Implementation Plan...</td>
<td>Prior to installation of this design change, it must be determined that the installation relationship between this design change and any other previously installed design change(s) or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>21/07/2015</td>
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<td>10054157</td>
<td>SABENA TECHNICS BGC</td>
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<td>EC 155 B, EC 155 B1</td>
<td>ADS-B OUT Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product have been amended and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement N ‘ 2037-01-A-2301- R00 or later revisions of the above listed documents approved by EASA Master Data List N° 2037-01-A-2103- R01/COMPLIANCE RECORDS N° 2037-01-A- 2104- R01INSTRUCTIONS FOR CONTINUED AIRWORTHINESS N° 2037-01-A-2211-R01</td>
<td>Prior to installation of this design change, it must be determined that the installation relationship between this design change and any other previously installed design change(s) or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>22/07/2015</td>
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List of EASA Supplemental Type Certificates

described in the Certification Plan, Doc. No. CP-0201-2500, Rev. C, dated 06.07.2015.

The Certification Basis (CB) for the original product as amended by the following paragraph(s) at a later amendment:
- CS 25 amdt 4 - CS 25.1, CS 25.1301, CS 25.1309, CS 25.1501, CS 25.1529, CS 25.1581. The requirements for
environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Approval valid only for MSN 37560 - The installation of the A-00-12/49-ICA section 4 - The airplane shall be operated
- supplementary to Installation and Approval of the above listed documents approved by EASA. 
- The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment:
- CS 25.1301, CS 25.1309, CS 25.1501, CS 25.1529, CS 25.1581. The requirements for
environmental protection and the associated certified noise and/or emissions levels of the original product.

EASA STC 10041604 (EASA Project Number 0010008975-001) is applicable to this certificate/ approval. Prior to installation of this change/ repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce radiative effect upon the airworthiness of the product.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce radiative effect upon the airworthiness of the product. 

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce radiative effect upon the airworthiness of the product. 

Airbus A300-203(‑1, +1) TCAS Change 7.1 Upgrade

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

Applelicable to aircraft serial number 255 only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce radiative effect upon the airworthiness of the product.

Applicable to aircraft serial number 255 only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce radiative effect upon the airworthiness of the product.

Applicable to aircraft serial number 255 only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce radiative effect upon the airworthiness of the product.
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<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<td>10054195</td>
<td>ADAC LUFTFAHRT TECHNIK GmbH</td>
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<td>MiBB-117 D-2</td>
<td>Installation of a Honeywell RDR-2000 Weather Radar NEW-PAGE</td>
<td>The Certification Basis for the original product as amendability the following additional or alternative airworthiness requirements: CS-29.Amtl. 2 (not required by 23 A.10121(4)&lt;(&gt;) - 29H&lt;(&gt;)&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(+)&lt;(+)&lt;(&gt;&lt;)+&lt;(+)&lt;(&gt;&lt;)+&lt;(+)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Installation of IFE server Rev. 1: Modification of system operational limitation. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification limitations are replaced by inspections. The requirements for structural soundness with respect to corrosion were addressed by inspection cycles. The lifetime limitation addressed by inspections is replaced by intervals. The LED modules which require less power are available. These lights are for general illumination and provide no emergency lighting. Prior to installation of this design change it must be determined that the interrelationship between the design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Installation of a Bruce Aerospace Cabin Lighting System. The STC installs new LED lighting systems which require less power. The lights are for general illumination and provide no emergency lighting. They require power and interface with a new control panel to allow operation. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Replacement of the lifetime limitation of 35 calendar years by periodic inspections. Structural soundness with respect to corrosion was addressed by tightening of 35 years. The limitation is replaced by inspections outlined in SB IS-28B2/E0-26 at short intervals. The lifetime limitation addressed by inspections is replaced by intervals. The requirements for structural soundness with respect to corrosion were addressed by inspection cycles. The requirements for environmental protection and the associated certification limitations are replaced by inspections. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification limitations are replaced by inspections. Aircraft Flight and Maintenance Manual issue 2A, replacement of pages 2.2.D, 2.3.C, and 23.2.D at revision 17, dated May 2015, Aircraft Flight and Maintenance Manual issue 3, replacement of pages 2.4.C, 3.2.1.C, and 6.10.F at revision 41, dated May 2015. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Replacement of the lifetime limitation of 35 calendar years by periodic inspections. Structural soundness with respect to corrosion was addressed by tightening of 35 years. The limitation is replaced by inspections outlined in SB IS-28B2/E0-26 at short intervals. Aircraft Flight and Maintenance Manual issue 2A, replacement of pages 2.2.D, 2.3.C, and 23.2.D at revision 17, dated May 2015, Aircraft Flight and Maintenance Manual issue 3, replacement of pages 2.4.C, 3.2.1.C, and 6.10.F at revision 41, dated May 2015. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Installation of a new LED lighting system. This STC installs new LED lighting systems which require less power. The lights are for general illumination and provide no emergency lighting. The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification limitations are replaced by inspections. Structural soundness with respect to corrosion was addressed by inspection cycles. Aircraft Flight and Maintenance Manual issue 2A, replacement of pages 2.2.D, 2.3.C, and 23.2.D at revision 17, dated May 2015, Aircraft Flight and Maintenance Manual issue 3, replacement of pages 2.4.C, 3.2.1.C, and 6.10.F at revision 41, dated May 2015. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.
<table>
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<tr>
<th>STC Holder</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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</thead>
<tbody>
<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>A321-231, A321-232, A321-233</td>
<td>Installation of a Boeing Aerospace Cabin lighting system. This STC installs new cabin lighting, ceiling and sidewall, with LED modules which require less power. These lights are for general illumination and provide no emergency lighting. They require power and interface with a new control panel at light...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Delta Engineering Master Data List, ref: 0704-0001-2372 Revision E, dated 20 April 2015</td>
<td>None</td>
<td>05/08/2019</td>
<td>Active</td>
<td>1054304</td>
</tr>
<tr>
<td>365 AEROSPACE DDS LIMITED</td>
<td>757-200</td>
<td>Installation of a Controller Pilot Data Link Communications (CPDLC) System Rev. 1.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 1525.301(c) &lt;p&gt;25.305(c) &lt;p&gt;25.307(c) &lt;p&gt;25.385(c) &lt;p&gt;25.386(c) &lt;p&gt;25.571(c) &lt;p&gt;25.581(c) &lt;p&gt;25.601(c) &lt;p&gt;25.603(c) &lt;p&gt;25.605(c) &lt;p&gt;25.607(c) &lt;p&gt;25.611(c) &lt;p&gt;25.613(c) &lt;p&gt;25.615(c)</td>
<td>1. Boeing 757-200 aircraft must have skin configuration 143N121-14 and GPS antenna location as defined by Boeing drawing 143N121-1. Each Boeing 757-200 aircraft, by serial number, must have a specific declaration of Design and Performance document calling up the necessary Amteq documents.</td>
<td>05/08/2019</td>
<td>Active</td>
<td>1054305</td>
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<td>NORTHWEST AEROSPACE TECHNOLOGIESI NC.</td>
<td>A340-313</td>
<td>Cabin Interior Reconfiguration of Lufthansa Airlines Aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Summary, Northwest Aerospace Technologies, Inc. document No. N4098-D06, Revision -, dated 2 August 2015 - Instructions for Continued Airworthiness, Northwest Aerospace Technologies, Inc. document No. N4098-D07, Revision A, dated 5 June 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/08/2015</td>
<td>Active</td>
<td>1054306</td>
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<tr>
<td>TENENCIA Ltd.</td>
<td>BAE 146-200, BAE 146-100, AVPRO 146-RJ85, AVPRO 146-RJ70, AVPRO 146-RJ15, AVPRO 146-RJ100</td>
<td>TCAS II Version 7.1 Upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Tenencia Ltd Modification TSN 794 issue 1 and all its referenced documents.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/08/2015</td>
<td>Active</td>
<td>1054313</td>
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<tr>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>LEARJET 31A</td>
<td>Upgrade TCAS II to ACAS II Change 7.1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Airplane Flight Manual Supplement, LR13-15003-050-MA-0, Revision 0, dated 17 July 2015. 2. Instruction for Continued Airworthiness, LR13-15003-050-MA-0, Revision 0, dated 17 July 2015. 3. Master Document List, LR13-15003-050-MA-0, Revision 0, dated 30 July 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/01/2022</td>
<td>Active</td>
<td>1054319</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>BOMBARDIER Inc.</td>
<td>10054346</td>
<td>ST00354BO</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Challenger 604 variant – Installation of SAT AFIS Antenna</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- ACS LS Modification Data Summary : EMR2000 Rev A - Mandatory Maintenance Manual Supplement. - DMM 5340-4001-8MMMS Rev NC - Instruction for Continued Airworthiness - DC-5340-4001-8FAA Rev NC</td>
<td>Applicable only For CS/604 Aircraft Serial No. 6528 and 5630. Refer to Limitation section of the approved Mandatory Maintenance ManualSupplement (MMMS).</td>
<td>08/02/2015</td>
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<tr>
<td>INNOVATIVE SOLUTIONS &amp; SUPPORT, Inc.</td>
<td>10054351</td>
<td>REV. 4</td>
<td>FAA ST00354BO</td>
<td>757-200, 757-300 Installation of an Innovative Solutions &amp; Support (IIS-I+S) Integrated Standby Unit (ISU), Part Number 80-80190-7, loaded with installed software</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments:CS-25 Amendment 17 pertaining to 25.1 (b), 25.21(a1), 25.29, 25.30(a), 25.33, 25.35(a), 25.37(a), 25.39(a)(3)(c)(d), 25.61(a)(x)(3) (i)(ii), 4.</td>
<td>- Aircraft Flight Manual Supplement (AFMS) Document ID-12310, Revision 3, dated 10 February 2021 - S Master Document List (MDL) Document ID-12303, Revision 5 dated 08 February 2021</td>
<td>- Refer to FAA Advisory Circular (AC) 20-188 for compatibility of this design change with previously approved modifications. - IS&lt;&gt;&amp;&lt;&gt;S ISU, Part Number 80-80190-7, loaded with top level software 7H-80190-33, is defined as: Article Part Number,...... Article Description 00-80190-T 3...</td>
<td>23/03/2021</td>
</tr>
<tr>
<td>JET AVIATION AG</td>
<td>10054353</td>
<td>757-200, 757-200, 757-200 Installation of a new IES IA 500</td>
<td>MYSTERE-FALCON 50</td>
<td>ACAS II upgrade to software version 10.12 onwards: TPA-81A P/N 068-50000-2520 or TPA-10A P/N N960-0305450-001 from Honeywell TPA-10B P/N 940-0351/0451-010</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>AFM supplement Document ID: JBSL-M-00466AFM-01 Issue 1 dated 16.07.2015 or later revisions of the above listed documents approved by EASA MASTER DATA LIST Doc. No. JBSL-M-00466ML-DL-01 Issue 1 dated 06.08.2015</td>
<td>MF800 &amp; MF50 aircraft with prerequisite:ACAS II compliant TCAS, Honeywell TPA-81A or TPA-10A and EHS and/ or ENS-B compliant SSR ATC transponders and altitude data not provided through Gray (Gilham) coding.Prior to installation of this design change it must be determined that the interrelationship be...</td>
<td>07/08/2015</td>
</tr>
<tr>
<td>PMV ENGINEERING</td>
<td>10054384</td>
<td>A340-311-312-313</td>
<td>In Flight Entertainment (IFE) system Installation and Cabin/interior modification on A340. This modification deals with the installation of the Zodiac Infight Innovations IFE system(Rave) and interior upgrade (rearrangement of/seat and replacement of Y/C seats) on A340aircraft.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:JAR25 Change 15 (Refer to PMV document. Ref: SCA-0300-00R2 for the details/affected requirements)The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>PMV Master Document List, ref: SCA-00300 Rev. 01, dated 06 August 2018PMV Instruction for Continued Airworthiness - associated to A340 ZII IFE Installation, ref: SCA-00300-ICA, dated 21May 2015-associated to A340 Cabin modification, ref: CAB-00500-ICA, dated 01January 2015-associated to A340 SICMA M...</td>
<td>Prior to installation of this design change It must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a risk on the airworthiness of the product.</td>
<td>07/06/2015</td>
<td>Active</td>
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<tr>
<td>HELIFAB Inc.</td>
<td>10054379</td>
<td>FAA STC SR09580RC</td>
<td>S-76A, S-76B, S-76C, S-76D Installation of an aft cabin area Storage in accordance with Helifab Inc. installation drawings document list HF: ST76-ix-2550-002, Revision:R, dated January22, 2014.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- Rotorcraft Flight Manual Supplement, Helifab Inc. Document No HF-ST76- MD-2550-002, Rev. IR, dated September 14, 2012 - Installation drawing and documents list HF: ST76-ix-2550-002, Rev E, dated January 22, 2014 or later revisions of the above listed documents approved by EASA in accordance with EASA...</td>
<td>S-76 D through S/N 761036Prior to installation of this design change It must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a risk on the airworthiness of the product.</td>
<td>10/06/2015</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>EASA STC no</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>HELI-ONE</td>
<td>CPI-503 Automatic Deploiable ELT Installation NEW PAGE</td>
<td>STC Data List ref. ASI-SDL-0067-02, dated 07/08/2015 for revisions of the above listed documents approved by EASA.</td>
<td>10054383</td>
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<td>Active</td>
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<tr>
<td>ASI INNOVATION</td>
<td>Embracer 170-100 Cabin Modification (dual class 68 passengers)</td>
<td>STC Data List ref. ASI-SDL-0067-02, dated 07/08/2015 for revisions of the above listed documents approved by EASA.</td>
<td>10054382</td>
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<tr>
<td>PMV ENGINEERING</td>
<td>Venus IFE installation on A319/320</td>
<td>Limited to MSN 2592/ASA STC IM.AS.01793 (validated FAA STC/ST105468C) needs to be installed prior to this STC. Prior to installation of this design change it must be determined that the installation will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>10054395</td>
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<td>Active</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>TCAS 7.1 system upgrade on ATR 42 &amp; ATR 72/212A ATR 42/230/-400/-500</td>
<td>This STC is not valid for the ATR 42/500 '900 Version' and ATR 72/212A '600 Version' (refer to EASA TCDS EASA.A/08). Prior to installation of this design change it must be determined that the installation will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>10054396</td>
<td>REV.</td>
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<td>Active</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC Ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10054416</td>
<td>AEROTE &amp; CONCEPT</td>
<td>REV. 2</td>
<td>A320-200 AND A321-200</td>
<td>Retractable Hardpoints installation on airborne wireless platform activation on A320/A321 Rev.2</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: paragraph (c)(1) of a later amendment: CS25 at amdt 10, 15.25.1301, 15.1309, 25.1563, 25.1431, 25.1529, 25.1581, 25.1585. The requirements for environmental protection...</td>
<td>Master Data List AE14263MDL rev. 3 dated 23 Mar 2016</td>
<td>As per AFMS: &quot;'WiFi' system is prohibited to work and must be switched OFF during refueling and Low Visibility Approaches. Use &quot;WiFi ON&quot; switch located on CCR in the FW of the first R/H Overhead Bin and &quot;TPEDs including GSM personal devices are prohibited to work and must be sw...</td>
<td>21/04/2016</td>
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<tr>
<td>10054417</td>
<td>HARTZELL PROPELLER INC.</td>
<td>REV. 1</td>
<td>FAA STC SE03232CH</td>
<td>Engine Preheat Kit - EASA Validation of FAA STC SE03232CH Installation of Engine Preheat Kit, p/n TSPA4CLY, 2953.115 for 115 volt application Rev. 1: changes the EASA Design Approval Holder's name and address from that for Tanis Aircraft Products (&quot;Tanis&quot;) to those for Hartzell.</td>
<td>The Certification Basis for the original product remains applicable to the new certification basis for the new product.</td>
<td>Engine Preheat Kit drawing TSG25-1 Revision F, dated 8 March 2023 (+); &gt; Preheat Kit Installation Instructions TP2925 Revision E, dated 12 April 2019 (+); &gt; Instructions for Continued Airworthiness (ICA) document TCA 1500 Revision D, dated 7 August 2023, and Operating Guide document TPG 1500 Revision...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the new product. This STC is valid for engine types and models...</td>
<td>17/01/2024</td>
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<tr>
<td>10054420</td>
<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINE S.A.R.L.</td>
<td>PA-32-300 (CHEROKEE SIX 300)</td>
<td>Engine Preheat Kit - EASA Validation of FAA STC SE03232CH Installation of Engine Preheat Kit, p/n TSPA4CLY, 2953.115 for 115 volt application Rev. 1: changes the EASA Design Approval Holder's name and address from that for Tanis Aircraft Products (&quot;Tanis&quot;) to those for Hartzell.</td>
<td>Engine Preheat Kit installation on PA32-300 serial number 4481/14 Revision 1 dated 04/05/2015 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the new product.</td>
<td>12/08/2015</td>
<td>Active</td>
<td>10054420</td>
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<tr>
<td>10054422</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SR0666IRC</td>
<td>Installation of Retractable Hardpoints System</td>
<td>Installation of Retractable Hardpoints System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>13/08/2015</td>
<td>Active</td>
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<tr>
<td>10054423</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>MD-11, MD-11F</td>
<td>TCAS Change 7: Replacement of the TCAS II computer with software version 7.0 currently installed on Lufthansa German Airlines (DLH) MD11 aircraft by a new-generation TCAS II computer with software version 7.1.</td>
<td>TCAS Change 7: Replacement of the TCAS II computer with software version 7.0 currently installed on Lufthansa German Airlines (DLH) MD11 aircraft by a new-generation TCAS II computer with software version 7.1.</td>
<td></td>
<td>19/02/2015</td>
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<td>STC Holder</td>
<td>STC ref.</td>
<td>Rev. level</td>
<td>Foreign/NA A STC ref.</td>
<td>Model</td>
<td>PLM ref.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>SCANDINAVIAN</td>
<td>ATR72-212/212A</td>
<td>00</td>
<td>ATR42-300/320/400/500</td>
<td>ATR airplane – Installation of provision for portable Electronic Flight Bag (EFB) with wireless connectivity and optional surveillance camerasystem. The installation provides, as a baseline, the provision for Portable EFB, FTE Displays in the cockpit on both pilot sides, including two windows/mounts, etc...</td>
<td>ATR42/444</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 4, for all applicable requirements, as an option to comply. These requirements for environmental protection and the associated certification noise and/or emissions...</td>
<td>Aerospace Flight Manual Supplement for Portable EFB, ScandanavianAvionics document 2084240-AM A issue 2084240-AM B issue 17</td>
<td>1. Only products that meet the Aircraft configuration specified in ScandanavianAvionics Engineering Order 2084240-E0 (chapter 4) can be modified with this STC.</td>
</tr>
<tr>
<td>HARTZELL PROPELLER</td>
<td>REV. 1</td>
<td>SEE FAA AML</td>
<td>SEE FAA AML</td>
<td>A330-243</td>
<td>Installation of Engine Preheat Kit, p/n TA2927-1 for 115 volt application, in accordance with Engine Preheat Kit drawing TA2927-1, Revision G, dated 6 March 2013, and Preheat Kit Installation Instruction TNP2927, Revision G, dated 12 April 2018, installation of Engine Preheat Kit, p/n TA2927-2 for...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 14 -25.301, 25.303, 25.305, 25.307, 25.61, 25.611, 25.613, 25.619, 25.625, 25.896, 25.1301...</td>
<td>Instructions for Continued Airworthiness (ICA) document TCA1000, Revision C, dated 5 October 2022, and Operating Guide document TPG1000, Revision E, dated 29 October 2020, installation of Engine Preheat Kit, p/n TA2927-1 for...</td>
<td>Prior to installation of this change/repair it must be determined if there is an adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>AMES - AEROSPACE AND</td>
<td>REV. 1</td>
<td>SEE FAA AML</td>
<td>SEE FAA AML</td>
<td>A330-243</td>
<td>Installation of Engine Preheat Kit, p/n TA2927-1 for 115 volt application, in accordance with Engine Preheat Kit drawing TA2927-1, Revision G, dated 6 March 2013, and Preheat Kit Installation Instruction TNP2927, Revision G, dated 12 April 2018, installation of Engine Preheat Kit, p/n TA2927-2 for...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 14 -25.301, 25.303, 25.305, 25.307, 25.61, 25.611, 25.613, 25.619, 25.625, 25.896, 25.1301...</td>
<td>Instructions for Continued Airworthiness (ICA) document TCA1000, Revision C, dated 5 October 2022, and Operating Guide document TPG1000, Revision E, dated 29 October 2020, installation of Engine Preheat Kit, p/n TA2927-1 for...</td>
<td>Prior to installation of this change/repair it must be determined if there is an adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>ALAMO ENGINEERING</td>
<td>REV. 2</td>
<td>CL-600-2E2</td>
<td>CL-600-2E2 FJ1907</td>
<td>CL-600-2E2</td>
<td>TCAS Change 7.1 Upgrade Rev. 2 transfers the STC from “Deutsche Aircraft GmbH” to “Alamo Engineering GmbH”.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 Amendment 4, for all applicable requirements, as an option to comply. These requirements for environmental protection and the associated certification noise and/or emissions levels of the original product airworthiness and remain applicable to this certificate/approval.</td>
<td>328 Design GmbH Aircraft Flight Manual Supplement No HBKAFMS101/144200 01 Issue A (or later revisions of the above listed documents approved by EASA).</td>
<td>1. This modification must be installed in conjunction with the following references: a) Airbus Modification 20504/D03138 (b) EASA STC No. 10015507 rev 2c) EASA approval of Project No. 001037133 (EASA Validation of TAP/Project no. ST1473/ATA.7.2 Monitors and components of the IF System, including In...</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>1005444</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
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<td>ATP</td>
<td>ATP aircraft - Provision of dual portable Electronic FlightBag (EBF). The installation of the EBF provision comprises a22 window mounting of a swivel and a tablet holder, 2 EFB control panels, immediate integration centre and 2 CII after 28VDC to the EBF systems.</td>
<td>The Certification Basis for the original product as amended/ the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment/Add elect to comply, CS-25 amendment 4, for all applicable requirements, except for 25.341 and 25.561 as documented by JAR...</td>
<td>AFS01 Flight Manual Supplement for Dual Portable Electronic FlightBag (EBF) Provision, Scandinavian Avionics document 201712-A1, issued- Instructions for Continued Airworthiness for Provision for dualportable EBF, Scandinavian Avionics document 201512-C4, issue for later revisions of the ab...</td>
<td>17/02/2015</td>
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<tr>
<td>1005449</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 1</td>
<td>1900D</td>
<td>P-NAV &amp; LPV equipment installation in which 1900D.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Installation of AEROTEC modification No 15071 in accordance with AEROTEC Master Drawing List ref AE15071MDL, Rev 4, dated 18/12/2015. Operate in accordance with AEROTEC Flight Manual Supplement ref AE15071AFMS, Rev 3, dated 15/11/2017. Instructions for Continued Airworthiness in accordance with AEROTEC.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/12/2017</td>
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<tr>
<td>1005450</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC</td>
<td>SAD1760W</td>
<td>EMB-505</td>
<td>Installation of an Aircraft Universal Cabin System (UCS) 5000 with/without local Area Network (WLAN).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Duncan Aviation, Inc, Master Document List Document No CERTDOC-000000179, Revision A, dated 01 August 2014. Operate in accordance with Duncan Aviation, Inc, Flight Manual Supplement, Document No EDOC-0000017904, Revision A, dated 01 August 2014 or later revisions of the...</td>
<td>This Design Change is only applicable to Embraer EMB-505 aircraft without/with Aircraft Cockpit Biz system installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will...</td>
</tr>
<tr>
<td>1005451</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC</td>
<td>SAD1760W</td>
<td>EMB-505</td>
<td>Installation of “Quasar II” Cabin Lighting with eConnect and NoseLighting.</td>
<td>The Certification Basis (CB) for the original product as amended/ the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment FAR 23.856 Am 23-62 (the requirements for environmental protection and the associated certified noise and/or emissions levels)...</td>
<td>Installation in accordance with Duncan Aviation Master Document List ref EDOC-018333, Rev A, dated 22 June 2015. Maintain in accordance with Duncan Aviation Instructions for Continued Airworthiness (ICA) ref EDOC-018321, Rev A, or at later revisions of the above listed documents approved by EASA in accordance...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>1005475</td>
<td>ECMS AVIATION SYSTEMS GMBH</td>
<td>REV. 3</td>
<td></td>
<td>EAD03</td>
<td>EAD03 External loads system for NHEC and HEIC operations/Revision 2.- Introduction of a shelf life definition in the Airworthiness/Imitation Section of the Instructions for Continued Airworthiness (Component maintenance manual) - Minor editorial corrections of the Component maintenance manual Reason fo...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unaltered and remain applicable to this certificate/approval without any im...</td>
<td>Initial issue - Flight Manual Supplement EAD03, RFSMS, EASA, Rev. 0 dated 1 May 2015 - Airworthiness limitation section as per the Component Maintenance Manual Chapter 11, Original Issue dated 09th August 2015- Equipment and configuration list EAD03 ECL, Rev. 1 dated 11th August 2015- List of certificatio...</td>
<td>No Personnel Carrying Device System (PCDS) is considered without this approval A separate airworthiness approval for PCDS is needed for HEC operating complex PCDS Installation of the EAD03 external load system to a MBB-BK117 D-3 and D-3M using RA04 /PA04 adapters is only possible if D-2 style f...</td>
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<td>STC Holder</td>
<td>Active Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 1</td>
<td>10054487 10054490 10054506 112TCA 114TC 114B 114C 114D 114TC 114D 114C 114D</td>
<td>Installation of 3-blade constant speed propeller MTV-0-D200-58 on Textron and/ or AMT Air Corp. in accordance with AMT-Propeller Master Documentation List No. E-2527, Rev 1; 1T to be operated in accordance with AMT-Propeller AFM-S Document No. E-2667, Rev 1; or later revisions of the associated documents approved by the FAA or EASA to be installed i.a.w.</td>
<td>According to: MT-Propeller Master Documentation List No. E-2667, Rev 1. To be operated in accordance with AMT-Propeller AFM-S Document No. E-2667, Rev 1.</td>
<td>06/02/2018</td>
<td>Active</td>
<td>10054487</td>
<td></td>
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<tr>
<td>MOTORFLUG BADEN-BADEN</td>
<td>REV. 1</td>
<td>10054506 228-200-2011-202 228-200-100-101</td>
<td>Approval of Motorflug Modification ref 28145F181335-0109 to provide ADS-B capability.</td>
<td>CS-ACNS Initial Issue, Subpart D. The requirements for environmental protection and the associated capability of the aircraft to maintain ADS-B capability.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/01/2019</td>
<td>Active</td>
<td>10054506</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10054518</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERINGCO RPRATION AG (AIEC)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. AIEC AG AIRPLANE FLIGHT MANUAL SUPPLEMENT NO. ACS362-071 REVISION 1 DATED 14 AUGUST 2015</td>
<td>ACS362-071 REVISION 1 DATED 14 AUGUST 2015</td>
<td>RESTRICTED TO ASN 5042</td>
<td>24/02/2015</td>
<td>Active</td>
<td>10054518</td>
<td></td>
</tr>
<tr>
<td>10054539</td>
<td>TDG AEROSPACE INC</td>
<td>Universal fault interrupter/install/Electrical protection device/thatInterrupts and electrical source immediately afterdetecting a ground, aile-to-line or an arc fault(s).</td>
<td>The Certification Basis for the original product as amended/this following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment 25.981(a)(3) A (d) Amdt: 25-125 The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>UFI MD-80 Installation kit Master Drawing List Doc. 10-3210-0002 Rev. EAFM Supplement 20-3210-0002 dated 4 March 2014 Instruction for Continued Airworthiness (ICA) 20-3210-0001 Rev. A</td>
<td>See doc. 20-3210-0002 dated 4 March 2014 &quot;FAA approved AFS Supplement&quot; Sect. 1</td>
<td>25/08/2015</td>
<td>Active</td>
<td>10054539</td>
</tr>
<tr>
<td>10054572</td>
<td>PMV ENGINEERING</td>
<td>TCAS 7.1 upgrade on B777</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. - Master Document List Avi-AE-2009-MDL-R003 dated 13 Aug 2015. Aircraft Flight Manual Supplement Avi-AE-2009-MDL-R004 dated 28 Jul 2015</td>
<td>AMS-2015-0015 Instructions for Continued Airworthiness Avi-AE-2009-MDL-R005 dated 28 Jul 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change(s) and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/08/2015</td>
<td>Active</td>
<td>10054572</td>
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<tr>
<td>10054574</td>
<td>TELEDYNE UK LIMITED</td>
<td>Installation of WGL-FI System on Airbus A319/320 Series Aircraft Rev. 1. Transfer from Teledyne Limited to Teledyne UK Limited.</td>
<td>The Certification Basis for the original product as amended/this following additional or alternative airworthiness requirements - Special Condition (c): ORI-F-02, Network Security. - The following paragraph(s) at a later amendment 25.981(a)(3) A (d) Amdt: 25-125 The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Teledyne Controls Flight Manual Supplement MOD-FMS-033 Issue 01 dated 05 May 2015 Teledyne Controls Master Data List MOD-MDL-033 Issue 02 dated 11 August 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change(s) and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/08/2015</td>
<td>Active</td>
<td>10054574</td>
</tr>
<tr>
<td>10054581</td>
<td>DEC TECHNOLOGIES, INC.</td>
<td>Installation of a locking hinge on the baggage door of the BELL 429 rotorcraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement Report No. DEC13052-RFS dated 2014</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change(s) and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/08/2015</td>
<td>Active</td>
<td>10054581</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>SJT AEROSPACE CORPORATION</td>
<td>STC ref. ST02340CH</td>
<td>SJT Aerospace Corporation Instructions for Continued Airworthiness:</td>
<td>SJT Aircraft Systems Corporation Instructions for Continued Airworthiness: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product were unchanged and remain applicable to this certificate/</td>
<td>27/02/2015</td>
<td>Active</td>
<td>10054582</td>
<td></td>
<td></td>
</tr>
<tr>
<td>AERODATA AG</td>
<td>Installation of AD-AFIS-0255 into B300.</td>
<td>SJT Aircraft Systems Corporation Instructions for Continued Airworthiness: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product were unchanged and remain applicable to this certificate/</td>
<td>Installation, Operation, and Maintenance according to the Master Drawing List document number MDL-AI-C-874 Rev. 1 dated 10th July 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>27/08/2015</td>
<td>Active</td>
<td>10054583</td>
<td></td>
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</tr>
<tr>
<td>Q.C.M. DESIGN GmbH</td>
<td>Hook Release Modification Installation of a switch to allow the pilot the option of selecting either the existing upper hook or a lower hook for cargo release:</td>
<td>SJT Aircraft Systems Corporation Instructions for Continued Airworthiness: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product were unchanged and remain applicable to this certificate/</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/08/2015</td>
<td>Active</td>
<td>10054584</td>
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<tr>
<td>SPECTRUM AEROMED, LLC</td>
<td>Air Ambulance Conversion Reason for revision 1: Change of ownership from Smoky Creek Enterprises, LLC to Spectrum Aeroned LLC or to Spectrum Aeroned:</td>
<td>SJT Aircraft Systems Corporation Instructions for Continued Airworthiness: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product were unchanged and remain applicable to this certificate/</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/04/2019</td>
<td>Active</td>
<td>10054587</td>
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<td></td>
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<tr>
<td>PRO STAR AVIATION, L.L.C.</td>
<td>Upgrade of the BendixKing CAS 87A TCAS II Change 7.0 to CAS 87B TCAS II:</td>
<td>INSTALLATION DATA LIST: Document Number 21238-IDL-101, Revision IR - EASA Approved Airplane Flight Manual Supplement:</td>
<td>An existing EASA-approved installation of a BendixKing CAS 87A TCAS II (Change 7.0) system with TCAS Processor Unit (TPU) part numbers identical to System Installation Manual 21238-SM-101. Prior to installation of this design change it must be determined that the relationship between these...</td>
<td>28/08/2015</td>
<td>Active</td>
<td>10054589</td>
<td></td>
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<tr>
<td>STC Holder</td>
<td>STC No.</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>1005461</td>
<td></td>
<td>TCA- SH14-31</td>
<td>EC 130 T2</td>
<td>Installation of Cargo Pods</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing AA-U001 Cargo Pods Installation: MSL-13-063 issue B0 later revisions of the above listed documents approved by FAA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement Instructions for Continued Airworthiness Cargo Pods InstallationCAA-AHCA-273 rev...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>SWA CERTIFICATION SERVICES Ltd.</td>
<td>1005467</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>A330-243</td>
<td>320Y LOPA Configuration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Configuration Scheme: 3135-MCS-01, Issue 2 Compliance Report, 3135-COM-01, Issue 2 for later revisions of the above listed documents approved by EASA.</td>
<td>Other: Applicable only for MSN 14981 under STC 10037490/Pinto installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>AVIATECH &amp; CONSULTING, SOCIEDAD LIMITADA</td>
<td>10054617</td>
<td>REV. 2</td>
<td>AT-802 AT-502 AT-502</td>
<td>INSTALLACION DE CONPUEARTAS ANTINCENDIOS &quot;Design and installation of compartmentalized gates for firefighting&quot; Rev. 2 Transfer from TRABAJOS AEREO ESPEJO S.L. to AVIATECH A CONSULTING SOCIEDAD LIMITADA</td>
<td>The Type and OSG Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval as: 1) Plan de Certificación PD01-12 Rev. 3, Fecha: 03/08/2015 (2) Planos Fotografías y Esquemas PL01-12 Rev. 3, Fecha: 17/07/2015 (3) Boletín de Ingeniería - Instrucciones de Instalación de los sistemas de detección de extinciones de incendios.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2022</td>
<td>Active</td>
</tr>
<tr>
<td>LUMA TECHNOLOGIES LLC</td>
<td>10054619</td>
<td>REV. 2</td>
<td>FAA STC SA02133SE</td>
<td>SEE FAA AML</td>
<td>Installation of Luma Technologies LT-4600 Series Integrated Lighting Emitting Diode (LED) Display Panel and LT-3500 LED light modules, and LT-2103 series landing gear annunciator Rev. 1: Correction of title and description Rev. 2: Correction of title and description Luma Technologies Master Drawing List (MDL) as listed on the AML or FAA approved revision. Luma Technologies Instructions for Continued Airworthiness (CA) as listed on the AML, or later FAA accepted revision.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval, except where amended by additional or later supplements as indicated on FAA/STC SA02133SE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/10/2020</td>
</tr>
<tr>
<td>LUFTHANSA TECHINIC AG</td>
<td>10054649</td>
<td></td>
<td>A330-300</td>
<td>New Premium Economy Class (A330-300) the main cabin of the main floor on the aircraft, the change consists of the removal of 10 per cent of the Economy Class Seats between door 2 and door 3 and the installation of new developed Premium Economy Seat with a Panasonic IFEC System instead. The new seating...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document STDOT25-139M issue 1 rev 0, Compliance/Rev List CR0025-139M issue 1 rev 0. Other: Applicable only for MSN 280, 280C, 280F, 280FX.</td>
<td>Other: Applicable only for MSN 280, 280C, 280F, 280FX.</td>
<td>According to the design related to this STC. This modification must be included in conjunction with: a) SB AIM A330-25-050 - Introduction of a three part class dissolution (Minor Change 2MM37-14b) STC 10045705 rev (or later revisions). b) Non Electric Floor PathMarking System, together with...</td>
</tr>
<tr>
<td>AIRWOLF AEROSPACE L.L.C.</td>
<td>10054670</td>
<td>REV. 1</td>
<td>FAA STC SR03485CH</td>
<td>F-28F-8, F-28A, F-28C, F-28C-2, F-28F 480, 480B 260, 260C, 260F, 260FX</td>
<td>Title: Installation of Airwolf Tension Torque Strap Assemblies: Installation of Airwolf Tension Torque Strap Assemblies PA AA-ECD-084-480 and PA-ECD-084-260 on Enstrom Helicopters by using Installation Instructions AA-ECD-084-260/480 Rev. A dated 9 June 2015 The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Airworthiness Limitations Section, Chapter 4 of the Continuing Airworthiness Maintenance Manual Tension Torque Strap Assemblies on Enstrom Helicopters Corporation as MRO: A 480 &amp; 480B Model ROTORcraft &amp; F-28/280 Model Series Rotorcraft Airwolf Aerospace LLC Report Number.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAAs</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>STC TWENTY ONE LIMITED</td>
<td></td>
<td>CA-A</td>
<td>BD-700-1A10</td>
<td>Introduction of Upgraded IFE system for STC Twenty One Flight Manual Supplement S21-FMS-012 Issue 1</td>
<td>The Certification Basis for the original product as amended by this supplemental or alternative airworthiness requirements (special condition) applies to the associated certified noise and/or emission levels of the original product.</td>
<td>STC Twenty One Flight Manual Supplement S21-FMS-012 Issue 1 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>07/02/2015</td>
</tr>
<tr>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td></td>
<td></td>
<td>F177RG</td>
<td>Installation of 3-blade constant speed propeller MTV-9-B/198-52</td>
<td>The Certification Basis (CB) of the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection and the associated certified noise and/or emission levels.</td>
<td>According to MT-Propeller Master Document List No. E-2659 the installation of a 3-blade constant speed propeller MTV-9-B/198-52 on Cessna 177A, 177B, 177RG, F177RG according to MT-Propeller AFM Document No. E-2659 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>07/02/2015</td>
</tr>
<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td></td>
<td></td>
<td>EC120 B</td>
<td>MOD 14250 – Installation of Garmin 500/500H/500W-PAGE</td>
<td>The Certification Basis (CB) of the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emission levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>RFID Supplement (/&gt;), EASA CAA FN-36Amdt.10</td>
<td>According to MT-Propeller Master Document List No. E-2659 the installation of Garmin 500/500H/500W-PAGE on EC120 B must be performed in accordance with EASA Certification Programme G500H installation EASA project n° 001000404D.</td>
<td>None</td>
</tr>
<tr>
<td>GLOBAL ENGINEERING AND TECHNOLOGY</td>
<td></td>
<td></td>
<td>FAA STC SA1788WM</td>
<td>Modification of Cabin Interior and Seating Configuration</td>
<td>The Certification Basis (CB) of the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emission levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>GETI Airplane Flight Manual Supplement No. AFMS120-101213, Rev. A, FAA approved January 28, 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>None</td>
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<tr>
<td>THE MONROE COMPANY, LLC</td>
<td></td>
<td>REV. 1</td>
<td>FAA ST02692CH</td>
<td>Installation of a Rockwell Collins IDS-3000 integrated cockpit flight displays system and other avionics modifications</td>
<td>The Certification Basis (CB) of the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emission levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List Great Lakes Aviation Holding document No. AE1000003 (&gt;); Revision H, dated 16 December 2011 Aircraft Flight Manual Supplement, Great Lakes Aviation Holding document No. AEN19007, Revision B, dated March 12, 2015 for later revisions of the above listed documents approved by EASA.</td>
<td>For aircraft equipped with Williams FJ-44-3AP engines, the engines must be installed in accordance with FAA STC ST0366/0 installed by EASA. These IDS-3000 is to be used for engine instrumentation display in accordance with this STC. Prior to installation of this design change it must be determined that...</td>
<td>None</td>
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<tr>
<td>HELI-ONE</td>
<td></td>
<td></td>
<td>TCCA STC HH11-38</td>
<td>Installation of Composite Utility Interior in accordance with Heli-One Master Drawing List CH1676-25-094 MDL</td>
<td>The Certification Basis (CB) of the original product remains applicable to this certificate/approval.</td>
<td>Master Drawing List CH1676-25-094 MDL, revision A heli-system installed by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>None</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10054711</td>
<td>HA AEROSPACE (UK) LTD.</td>
<td>CAAC UK RESPONSIBIL ITY</td>
<td>CAA STC 400</td>
<td>747-400</td>
<td>Installation of LED Wash Lights</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unamended and remain applicable to this certificate approval.</td>
<td>Certification Plan HICP20024 Issue 1, dated 20 April 2015</td>
<td>Prior to installation of this design change must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/eff affect upon the airworthiness of the product.</td>
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<tr>
<td>10054713</td>
<td>GOODRICH LIGHTING SYSTEMS</td>
<td>FAA ST197BAT</td>
<td>727-100, 737-200</td>
<td>Installation of Fixed Wingtip Halogen Navigational Lighting System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unamended and remain applicable to this certificate approval.</td>
<td>BF-Goodrich Aerospace Lighting Systems Master Drawing List BFGL-R_342.00(Rev A dated June 7(9) 2012</td>
<td>Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/eff affect upon the airworthiness of the product.</td>
<td>10/09/2015</td>
</tr>
<tr>
<td>10054714</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAAC UK RESPONSIBIL ITY</td>
<td>A321-272N-251NX-252NX-271NX, A321-271LN-271NX-272NX, A321-272N-251NX-272NX, A321-271LN-271NX-272NX</td>
<td>FlightLink System Installation Rev. 4 To add the A319/NEO, A320/NEO and A321/NEO models within the flexibility model list.Rev. 5 transferred from Tenencia Ltd. to STC Twenty One Ltd</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unamended and remain applicable to this certificate approval.</td>
<td>- Master Data List: TEN 779 Issue 26 dated 04.Feb.2020- Maintenance Manual Supplement: TEN 779-04 (D3110.04E) Issue 4, dated 06.Jan.2019. Flight Manual Supplement: TEN 77947 (TEN/320/2010) Issue 2 dated 06.Jul.2017. For later revisions of the above listed documents/approved/received under the EASA sys. - EMI/EMC Test Procedure reference TEN 779/11 must be performed and successfully completed.</td>
<td>Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/eff affect upon the airworthiness of the product.</td>
<td>14/11/2022</td>
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<td>10054715</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 1</td>
<td>A321-231</td>
<td>Vietnam Airlines - Cabin rearrangement Cabin seating rearrangement 118 “Zodiac” Y/C new seats</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unamended and remain applicable to this certificate approval.</td>
<td>- Change Approval Sheet CAS-A320-15-002 rev 2- for Continued Airworthiness(9&lt;,9&gt;)</td>
<td>Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/eff affect upon the airworthiness of the product.</td>
<td>08/03/2016</td>
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<td>10054724</td>
<td>ALPINE AEROTECH LP</td>
<td>TCCA STC 4513-38</td>
<td>429</td>
<td>Bear Paw Kit Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unamended and remain applicable to this certificate approval.</td>
<td>Alpine Aerotech Master Data List No AAL-490-014-001, Revision AAlpine Aerotech Instructions for Continued Airworthiness No.AAL-490-015-0701, Revision N09 for later revisions of the above listed documents approved by EASA inaccordance with the Technical Implementation Procedures of EU/Canada Bilateral A...</td>
<td>Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/eff affect upon the airworthiness of the product.</td>
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<td>NORTHWEST AEROSPACE TECHNOLOGIES  INC.</td>
<td>10054735</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIESIC.</td>
<td>10054735</td>
<td>Installation of Locking Fuel Cap, DEC</td>
<td>Certification Summary, Northwest Aerospace Technologies, Inc. documentNo. N0497-D00, Revision -, dated 6 September 2015, -Instructions for Continued Airworthiness, Northwest Aerospace Technologies, Inc. document No. N0497-D70, Revision -dated 1 July 2015</td>
<td>Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>11/02/2015</td>
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<td>HELI-ONE</td>
<td>10054737</td>
<td>HELICOPTER INC.</td>
<td>10054737</td>
<td>Installation of Egress Lighting System(HEELS)</td>
<td>Master Data List CHS76-53-044, Rev F</td>
<td>Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>11/09/2015</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>10054749</td>
<td>AMAC AEROSPACE TECHNOLOGIESI AG</td>
<td>10054749</td>
<td>Installation of an open cook top in the galley area of a VIP interior,CRI D-4308</td>
<td>Airplane Flight Manual Supplement: AFMS-DM01-2500 revision A dated26.08.2015,Instruction for Continued Airworthiness: ICA-DM01-2500 revision A dated07.07.2015,Master Minimum Equipment List Supplement MMELS-DM01-2500 revision A</td>
<td>Limited to S/N 5040Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>14/09/2015</td>
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<td>DEC TECHNOLOGIES, INC.</td>
<td>10054784</td>
<td>DEC TECHNOLOGIES, INC.</td>
<td>10054784</td>
<td>Installation of a Cabin Crew Manual CCM-DM01-2500 revision A</td>
<td>Master Data List MDL-DM01-2500, Rev. F, dated03/05/2015</td>
<td>Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noise/effect upon the airworthiness of the product...</td>
<td>16/09/2015</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>10054785</td>
<td>BELL HELICOPTER TECHTRON, INC</td>
<td>10054785</td>
<td>Installation of Egress Lighting System(HEELS)</td>
<td>Master Drawing List, Report No. AA-10059, Revision A, dated September 23, 2014,Flight Manual Supplement, ReportNo. AA-10065, No Revision, dated November 24, 2014,Instruction for Continued Airworthiness, Report No. AA-10064, NoRevision, dated August 28, 2014</td>
<td>Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/ or repair will introduce noise/effect upon the airworthiness of the product...</td>
<td>16/09/2015</td>
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### List of EASA Supplemental Type Certificates

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<th>STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA/Code</th>
<th>Model</th>
<th>Description</th>
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<tr>
<td>1005478</td>
<td>AVIONICARE Ltd</td>
<td>10054788</td>
<td>FAA STC ST01662CH-D</td>
<td>A330-223</td>
<td>Installation of Flight Data Acquisition Storage &amp; Transmission System (FAST)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>None</td>
<td>16/02/2015</td>
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<tr>
<td>1005479</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>10054797</td>
<td>FAA STC ST01662CH-D</td>
<td>A330-223</td>
<td>Installation of Flight Data Acquisition Storage &amp; Transmission System (FAST)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>None</td>
<td>20/10/2017</td>
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<td>1005480</td>
<td>STANDARD AERO</td>
<td>10054800</td>
<td>FAA STC ST01662CH-D</td>
<td>A330-223</td>
<td>Installation of Flight Data Acquisition Storage &amp; Transmission System (FAST)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>None</td>
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<td>1005481</td>
<td>AMTECH AERONAUTICAL LIMITED</td>
<td>10054809</td>
<td>FAA STC ST01662CH-D</td>
<td>A330-223</td>
<td>Installation of Flight Data Acquisition Storage &amp; Transmission System (FAST)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>None</td>
<td>17/09/2015</td>
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<td>1005482</td>
<td>B/E AEROSPACE, Inc.</td>
<td>10054810</td>
<td>FAA STC ST01662CH-D</td>
<td>A330-223</td>
<td>Installation of Flight Data Acquisition Storage &amp; Transmission System (FAST)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>None</td>
<td>18/09/2015</td>
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<td>1005483</td>
<td>PRATT &amp; WHITNEY ENGINE SERVICES, INC.</td>
<td>10054812</td>
<td>FAA STC ST01662CH-D</td>
<td>A330-223</td>
<td>Installation of Flight Data Acquisition Storage &amp; Transmission System (FAST)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
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<td>A330-243</td>
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<td>B747-8</td>
<td>Activation for GSM and WIFI. Installation. The Cabin Connectivity System is an end-to-end solution that provides GSM and WIFI services to aircraft passengers only. The system consists of hardware and software parts for the airborne GSM network to be connected to ground network via satellite, or any other.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>- JAR 25.820 Change 14JAR 25.435 Installation of angled seats &lt;br&gt; Limitations/Conditions: &lt;br&gt; 1) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/02/2015</td>
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<td>Boeing 787 TCAS V7.0 upgrade to V7.1</td>
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<td>787-400ER</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Certification Plan CP-0277 issue 1 Certification Plan CP-0278 issue 1 Compliance Record Report CRR-0277 issue 1 Compliance Record Report CRR-0278 issue 1 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/09/2015</td>
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<td>Virgin Australia Business Class Interior Reconfiguration/Installation of new business class seats and associated furniture – A330Virgin Australia cabin</td>
<td></td>
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<td>A330-243</td>
<td>The Certification Basis for the original product as amended the existing installation or alternative airworthiness requirements. Equivalent Safety Findings (CRI 4324-001 – Installation of angled seats) &lt;br&gt; Limitations/Conditions: &lt;br&gt; 1) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>MASTER DRAWING LIST JD-45AC02-1, rev AICA &amp; EASA for Continued Airworthiness JD-45AC04-1, rev 3 Certification Summary Report rev JD-45AC10-1, rev Newer later revisions of the above listed documents approved by EASA.</td>
<td>08/02/2017</td>
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<td>10054620</td>
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<td>MFG JET FALCON WESTFALIA</td>
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<td>A330-243</td>
<td>Certification Basis for the original product as amended the existing installation or alternative airworthiness requirements. Equivalent Safety Findings (CRI 4324-001 – Installation of angled seats) &lt;br&gt; Limitations/Conditions: &lt;br&gt; 1) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>-BOURNEMOUTH AVIATION (CONSULTANTS) LTD FLIGHT MANUAL SUPPLEMENT NO BAC-801-PAS81 ISSUE 1-BOURNEMOUTH AVIATION (CONSULTANTS) LTD MASTER DATA LIST NO BAC-851MDL01 ISSUE 5 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The approach approval shall fulfill the obligations of Part 21, 21/09/2015. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/or repair will introduce no adverse effect upon the airworthiness.</td>
<td>26/03/2021</td>
<td>Active</td>
<td>10054636</td>
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<td>Modification of VIP Interior installed in Airbus A330-243, MSN 1053.</td>
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<td>A330-243</td>
<td>Certification Basis for the original product as amended the existing installation or alternative airworthiness requirements. Equivalent Safety Findings (CRI 4324-001 – Installation of angled seats) &lt;br&gt; Limitations/Conditions: &lt;br&gt; 1) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>MDL ref 300700000 rev BICA ref 300705000 rev BCertification Plan ref 300700000, rev FFAA Statement Of Compliance dated on 23th Sept 2016AFM ref 300702000 rev AICA &amp; EASA for Continued Airworthiness (or subsequent revision).</td>
<td>This approval is limited to MSN 1053The following STC need to be installed, concurrently to this approval: STC 10032294, “Installation of Cabin Interior Noise Reduction Kit”. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/06/2015</td>
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<td>Wireless Fidelity (WiFi) Activation</td>
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<td>MF 50, MF 900</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>-3S Certification, Aircraft Flight Manual Supplement (AFM) 032-2255-A01-00, Revision IR, Dated: 08 December 2011 3S Certification, Aircraft Flight Manual Supplement (AFM)032-2255-A03-00, Revision IR, Dated: 25 September 2012 or later revisions of the above listed documents approved by EASA.</td>
<td>Applicability limited to MF50 and MF900 models. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/06/2015</td>
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<td>AEROTECH &amp; CONCEPT</td>
<td>Installation of Dual Indium/GPS Antenna</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, Aentric document No. AE14265MDL, Revision 0, dated 09 July 2015 - Instruction for Continued Airworthiness, Aentric document No. AE14265CA(I)&lt;/x&gt; - Revision 0, dated 09 July 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/02/2015</td>
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<td>LAKE CENTRAL AIR SERVICES</td>
<td>Installation of Synthetic Aperture Radar System (AKA SnowSAR)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List LCA01927-MDL, Rev. 1, TCCA approved January 28, 2013 Flight Manual Supplement LCA01927-CF, Rev. 2, TCCA approved February 4, 2013</td>
<td>Cessna 208B (EX) as defined in EASA TCDS IM.A.228, Section 2.2 are excluded Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/09/2015</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>Panasonic In-Flight Entertainment eX1 System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Delta Engineering Airplane Flight Manual Supplement No. 0664-09190-2373 Delta Engineering Instructions for Continued Airworthiness Document No. 0794-09191-2373</td>
<td>For configuration TL10792-01 Turkish Design Approval NG-2526001-DAO-D (raceways) is required.</td>
<td>30/09/2015</td>
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<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>Avion: Aero Data Rack Installation / With this major change No. STC-AT-090 a mission configuration for the AP68TP-600 “VIATOR”</td>
<td>The Certification Basis for the original product as amended following additional or alternative airworthiness requirements. The following paragraphs(x) at a later amendment: The following paragraphs of CS 23 Amendment 3 are applicable to the change: 23.1; 23.3; 23.21; 23.23; 23.26; 23.45; 23.49; 23.50</td>
<td>For aircraft specific to the change/repair it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair shall be incorporated prior to or in conjunction with the change. Prior to installation of this change/repair it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/11/2021</td>
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<td>HELI-ONE</td>
<td>Installation of EXIS Emergency Exit Lighting System in accordance with Heli-One Approval Control Document CHS76-33-022 ACD</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement EXIS Lighting System FMS CHS76-33-023, Revision0/0 or revisions above the listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/09/2015</td>
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<td>Model(s)</td>
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<td>10054913</td>
<td>TRANSPORTES AEREOS PORTUGUESES, S.ATAP MAINTENANCE &amp; ENGINEERING</td>
<td>A321-212, A321-213 A321-111, A321-112, A321-211 A320-211, A320-212, A320-214 A319-111, A319-115 A319-111, A319-112, A319-113</td>
<td>Installation of thrust reverser pivoting door actuator filling on CFM56-5B.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Report TAP-AC0136-CR-01R02 dated 18/09/2015/Compliance Check List TAP-AC0136-CCL-01R03 dated 19/05/2015 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>30/02/2015</td>
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<td>10054917</td>
<td>BOMBARDIER Inc.</td>
<td>REV. 1</td>
<td>TCJA C-LSA12-147D CL-600-2B16 (604 VARIANT)</td>
<td>Installation, Aircraft Completion (Special Mission Provision)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Modification Data Summary, ref. doc. RAZ-605-5898-MDS, Rev. B, dated 24/09 2013 - Modification Data Summary, ref. doc. RAZ-605-5898-MDS, Rev. B, dated 24/09 2013</td>
<td>Applicable Only For CL604 Aircraft Serial No. 5896 and 5898</td>
<td>01/10/2015</td>
<td>Active</td>
<td>10554917</td>
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</tr>
<tr>
<td>10054918</td>
<td>FEDEX EXPRESS CORP.</td>
<td>FAA ST03832AT</td>
<td>727-200</td>
<td>Installation of Nose Landing Gear Viewport with relocation of therewrap from aft of cargo bulkhead to forward of thercargo bulkhead.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment: JAR25 amdt 18: 25.301,25.303, 25.307, 25.307, 25.307(c) 25.57(b), 25.601, 25.603,25.605,25.607, 25.611, 25.613, 25.1301...</td>
<td>Federal Express Top Drawing B27-25-136 rev E dated Sept 6, 2009/AFM Supplement ER 08-121 dated Feb 10, 2010/Instructions for continued airworthiness 08-118 dated Aug 6, 2008/Later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent re...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/09/2015</td>
<td>Active</td>
<td>10554918</td>
<td></td>
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<tr>
<td>10054923</td>
<td>FEDEX EXPRESS CORP.</td>
<td>FAA ST01956AT</td>
<td>727-100, 727-200</td>
<td>Installation of a triple Rockwell Collins 619MSI VHF Transceivers and the placement of control heads with Gables D7406-04 to meet the 8.33kHz channel frequency spaces.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Federal Express Top Drawing B27-23-034 rev B dated May 10, 1999/AFM Supplement VHF-001 dated July 20, 1999/Later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/10/2015</td>
<td>Active</td>
<td>10554923</td>
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<tr>
<td>10054939</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 2</td>
<td>FAA ST0681W-I-D</td>
<td>SEE APPROVED MODEL LIST</td>
<td>TCAS II System Modification/Modification of a TCAS II System/Installation to Change 7 For Change?. Revision 2 of this STC includes revision to the AFM and administrative changes to the drawing index.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Refer to EASA APPROVED MODEL LIST for later revisions of the above listed documents/approved/accepted/owned by EASA in accordance with the Technical Implementation Procedures of EU/USA Bilateral Agreement.</td>
<td>EASA approved installation of a TCAS II system in a prerequisite to this STC. Refer to the Drawing Index for each model in the EASA AML Airplanes with a maximum operating altitude greater than 51000 feet at minimum authorized to upgrade from a TTR-920/CPN 622/697/5-521 to a TTR-92/for TTR-2100/Prior to in...</td>
<td>01/03/2019</td>
<td>Active</td>
<td>10554939</td>
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<tr>
<td>10054941</td>
<td>BUTLER AVIONICS, INC</td>
<td>FAA STC STO1791W</td>
<td>60</td>
<td>Installation of a Coltene AHC-3000A Series Attitude Heading Computer</td>
<td>The certification basis (CB) for the original products remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Butler National Corporation Master Drawing List No. 0286100, Rev. R, dated July 23, 2015 or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/11/2015</td>
<td>Active</td>
<td>10554941</td>
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<td>STC Holder</td>
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<td>Description</td>
<td>Rev.</td>
<td>Foreign/NA A STC ref.</td>
<td>Model</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated CERTIFIED Noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td></td>
<td></td>
<td></td>
<td>Falcon 1000A Aircraft Maintenance Manual or later revisions of the above listed documents approved by EASA Certification Plan Doc. No. JESL- M-00412R-01 Issue 4</td>
<td></td>
<td>02/10/2015</td>
<td>Active</td>
<td>1054970</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: Applicable regulations CT 25 Amendment 4 (95/1996 UK-CAA technical oversight receives approval).</td>
<td></td>
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<td></td>
<td>Master Data List. ref: 4802-01-A-MLD- R001Instruction for Continued Airworthiness, ref: 4802-01-A-ICA- R001 Aircraft Flight Manual Supplement. The modification is limited to the above listed documents approved by EASA.</td>
<td></td>
<td>02/10/2015</td>
<td>Active</td>
<td>1054960</td>
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<td>REV. 1</td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Deviation(s). 2/3/3-004 - Firm Handhold. The requirements for environmental protection and the associated CERTIFIED Noise and/or emissions levels of the original product are unchanged and remain applicable to 2x2...</td>
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<td>Modification Approval Sheet. ref: - S-5-2/3-004 Issue 4, dated 06-02-2015 Aircraft Flight Manual Supplement.</td>
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<td>06/10/2015</td>
<td>Active</td>
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<tr>
<td>FAA STO371NY</td>
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<td>Validation of FAA STC, Installation of Honeywell Primus Elite Avionics System.</td>
<td></td>
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<td></td>
<td>Applicable to Falcon 900EX (Falcon 900C version), Applicable to Falcon 900EX model. Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td></td>
<td>02/10/2015</td>
<td>Active</td>
<td>1054966</td>
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<tr>
<td>SCANNAVIA AERONAUTICA</td>
<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): SC F-9107-001, &quot;HIRF Protection&quot;. The following paragraph(s) at a later amendment: The Certification Basis for the parts changed or affected by the...</td>
<td></td>
<td></td>
<td></td>
<td>Rotorcraft Flight Manual Supplement 2091/141-AM A Issue 4, dated 09.08.2015 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/06/CF (or subsequent revisions thereof) and/or the Technical Implementation Procedures of EU USAbilateral Agreement Master Data...</td>
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<td>06/10/2015</td>
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<td>1054973</td>
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<td>AYESA AIR CONTROL</td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated CERTIFIED Noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td></td>
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<td>Airplane Flight Manual Supplement.</td>
<td></td>
<td>05/10/2015</td>
<td>Active</td>
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<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>05/10/2015</td>
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**List of EASA Supplemental Type Certificates**

- **STC Holder:** JET AVIATION AG, BASEL
- **Active:** Yes
- **Description:** The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated CERTIFIED Noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.
- **Rev.:**
- **Foreign/NA A STC ref.:**
- **Model:** MYSTERE-FALCON 900
- **Certification basis:** Falcon 1000A Aircraft Maintenance Manual or later revisions of the above listed documents approved by EASA Certification Plan Doc. No. JESL-M-00412R-01 Issue 4
- **Associated Technical Documentation:** Master Data List. ref: 4802-01-A-MLD-R001 Instruction for Continued Airworthiness, ref: 4802-01-A-ICA-R001 Aircraft Flight Manual Supplement. The modification is limited to the above listed documents approved by EASA.
- **Date of issue:** 02/10/2015
- **Status:** Active
- **System Record:** 1054970

**Notes:**
- The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated CERTIFIED Noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.
- The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Deviation(s). 2/3/3-004 - Firm Handhold. The requirements for environmental protection and the associated CERTIFIED Noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.
- The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): SC F-9107-001, "HIRF Protection". The following paragraph(s) at a later amendment: The Certification Basis for the parts changed or affected by the...
<table>
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<tr>
<th>STC Holder</th>
<th>FAA STC No.</th>
<th>Aircraft Model</th>
<th>Certification Basis</th>
<th>Description</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>CRANFIELD AEROSPACE LTD</td>
<td>CAA STC 3100-61A</td>
<td>JETSTREAM SERIES 3100</td>
<td>TCAS II Upgrade to Version 7.1</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-25 Amendment 1625.130125.13025.152525.158 The requirement for environmental protection and the associated certification not applicable to this certificate/approval.</td>
<td>None</td>
<td>05/11/2015</td>
<td>Active</td>
<td>10204970</td>
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<tr>
<td>ORCON CORPORATION</td>
<td>FAA STC ST09044A</td>
<td>767-200, 767-300</td>
<td>Blanket Insulation Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirement for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>05/11/2015</td>
<td>Active</td>
<td>10204997</td>
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<tr>
<td>ORCON CORPORATION</td>
<td>FAA STC STC0497LA</td>
<td>A318/A319/A320/A321 SERIES</td>
<td>Installation of Cabin Doors, Floor Panels and Forward Galleys Insulation Replacement Blanket</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirement for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA approved Orcon Corporation Index List FB3009 Revision B, dated August 17, 2005, or later FAA approved revision.</td>
<td>05/11/2015</td>
<td>Active</td>
<td>10204999</td>
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<tr>
<td>IAC - INDUSTRIAS AERONAUTICAS DECOMIRBA, Ltda.</td>
<td>FAA STC STC01824LA</td>
<td>U206G, TU206G, U206D, TU206D, U206E, TU206E, P206E, U206C, TU206C, 206H, 206H2, F206D, U206E</td>
<td>Static line parachutes support/Transit Support System. This support is manufactured and installed on the floor of the aircraft to provide structural integrity. It does not affect the loads carried by the seat belts. The support is designed to help the force dissipation to ensure the structural integrity.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): SC-C23-04/21 Parachuting Activity. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>AFM Supplement for Static Line Operation (IAC-C206) Special Airworthiness Approval DF 04-01 of the above listed documents approval by EASA Master Document List IAC-20655 [ML] Document Number F83006 Revision B dated January 29, 2014 or later FAA approved revision.</td>
<td>06/10/2015</td>
<td>Active</td>
<td>10205010</td>
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<tr>
<td>AVIONTEC AG</td>
<td>FAA STC STC01892LA</td>
<td>F260D</td>
<td>Parachutes Support System/Transit Support System. This support is manufactured and installed on the floor of the aircraft to provide structural integrity. It does not affect the loads carried by the seat belts. The support is designed to help the force dissipation to ensure the structural integrity.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): SC-C23-04/21 Parachuting Activity. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>AFMS Supplement for Static Line Operation (IAC-C206) Special Airworthiness Approval DF 04-01 of the above listed documents approval by EASA Master Document List IAC-20655 [ML] Document Number F83006 Revision B dated January 29, 2014 or later FAA approved revision.</td>
<td>06/10/2015</td>
<td>Active</td>
<td>10205011</td>
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<tr>
<td>ADS-B TECHNOLOGIES, L.L.C.</td>
<td>FAA STC STC02424AK</td>
<td>SEE FAA AML</td>
<td>Installation of the L-3 Communications NGT-10000002500 ADS-B system</td>
<td>The Certification Basis (CB) for the original product applicable to this certificate/approval. The requirement for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, L3-61-MLD 1, Revision 2, dated April 6, 2019.</td>
<td>11/12/2015</td>
<td>Active</td>
<td>10205012</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA (if nil)</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10055022</td>
<td>HELI-ONE</td>
<td>TC-A SH07-45</td>
<td>S-76A, S-76B, S-76C</td>
<td>Installation of Audio Panel NAT-380 in accordance with Avionics Design/Services Ltd. Master Drawing List MLD07124.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated airworthiness noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Configuration 1 and Co-pilot NAT-380 Audio Panel Installation-Master Drawing List MLD07124 Rev B Maintenance Manual Supplement H1C STC-76-23-0555-MMS Rev AConfiguration 2: Third NAT-380 Audio Panel Installation-Master Drawing List MLD07124 Rev B Maintenance Manual Supplement HCS/9293095-730.</td>
<td>06/10/2015</td>
<td>Active</td>
<td>10500002</td>
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<tr>
<td>10055024</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 1 FAA ST02928NY</td>
<td>777-200LR, 777-300ER</td>
<td>Installation of a Panasonic Avionics Corporation Global Communications Suite (GCS) Revision 1 removes the limitation for T-PED STC</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: following paragraphs(s) at a later amendment: CS25 amdr. 4: 25.381, 25.383, 25.889, 29.981, 25.1353, 25.1431, 25.1583, 25.1595, 25.1687. The requirements for environmental protection...</td>
<td>Delta Engineering Master Data List 0143-08223-2319 Revision O dated 24 June 2017 or later FAA approved revisions to Delta Engineering Master Data List previously dated: The Instructions for Continued Airworthiness and Airplane Flight Manual/Supplement as listed on the Master Data List.</td>
<td>09/10/2020</td>
<td>Active</td>
<td>10500024</td>
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<tr>
<td>10055027</td>
<td>ANORA INTERNATIONAL LLC</td>
<td>FAA STC ST04474LA</td>
<td>DC-9-81/82-83, MD-88</td>
<td>Installation of mechanical cargo loading system main deck installation of mechanical cargo loading system Pin 85130-1 0&lt;(&gt;,&lt;)&gt;-11 0&lt;(&gt;&amp;&lt;)&gt;-12 in accordance with FAA approved Ancra Int Controlled Data List OL86130 rev. C.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: following paragraphs(s) at a later amendment: CS25 amdr. 4: 25.381, 25.383, 25.889, 29.981, 25.1353, 25.1431, 25.1583, 25.1595, 25.1687. The requirements for environmental protection and the associated airworthiness noise and/or emissions levels of the original product are unchanged and remains...</td>
<td>Project Specific Certification Plan ANORA report 0674 rev. E. Documentation FAA approved Ancra International LLC Controlled data list OL85130 rev. C. or later revisions of the abovementioned documents approved by EASA in accordance with EASA EDD 2004/0454F (or subsequent revisions of this decision).</td>
<td>None</td>
<td>07/10/2015</td>
<td>Active</td>
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<tr>
<td>10055029</td>
<td>AERONAUTICAL ENGINEERS Inc.</td>
<td>FAA STC ST04343LA</td>
<td>DC-9-81/82-83, MD-88</td>
<td>MID-80 SFI/Conversion from passenger to co-pilot configuration including Class E Cargo compartment</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Conditions: CRI/SC D-02 Fire protection of Essential systems/equipment within Class E cargo compartments and CRI/SC D-03 Courrier compartment/Deviation (a) CRI D-01 Courrier T...</td>
<td>Project Specific Certification Plan doc. R-2817 Rev. N. Documentation FAA approved AFM supplement doc. R-847 Rev. B &amp; Airworthiness Limitation See: Maintenance Planning Document Supplement doc. R-867 IR Sect.Cargo Door Drawing List for the MD-80 Aircraft doc. R-831 rev. H withDept AES545 Rev. A (seven v...</td>
<td>See AFM supplement R-847 Rev. B/P for installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/10/2015</td>
<td>Active</td>
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<tr>
<td>10055030</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1 CAA UK RESPONSIBILITY</td>
<td>A340-642, A340-643</td>
<td>Installation of GoGo 2KU – Structural and Electrical Provisions.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 16–Various The requirements for environmental protection and the associated airworthiness noise and/or emissions...</td>
<td>STC Twenty One Ltd Instructions for Continued Airworthiness S21.TEC-047 3Issue 2 dated 10 May 2017, Section 3 Airworthiness Limitation (a) or later revisions of the above listed documents approved by EASA, STC Twenty One Ltd Master Document List S21.MDX-1238 Issue 10.</td>
<td>Airbus A340-642 and -643 aircraft, modified by this EASA STC, havedoublets and brackets attached to the external crown skin between frame PR 53.4 and FR 53.9 and stringers STR 4 L and STR 4 R. FAA EASA 2016-0152 (superseding EASA AD 2013-0127 for A340 aircraft) requires accomplishment of the actions...</td>
<td>13/06/2017</td>
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<td>10055031</td>
<td>AERODATA AG</td>
<td>REV. 1</td>
<td>1900D</td>
<td>Installation of AD-4F18-290 including additional RSBN receiver and the 50.04R Transponder</td>
<td>10055039</td>
<td>ADS-B TECHNOLOGIES, L.L.C.</td>
<td>REV. 2</td>
<td>FAA STC</td>
<td>Approval of the L-3 Communications NOT 9000/9000R ADS-B system</td>
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</table>

**List of EASA Supplemental Type Certificates**

| Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | 1. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 24/08/2016 | Active | 10055032 |
| --- | --- | --- | --- | --- | --- |
| Installation according to the Master Document List: MDL: AI/CA-865 Rev.1 dated 1st October 2015 and Service Bulletin SB-B1900-RUFS02-2154 Rev.0 Operation according to Aircraft Flight Manual Supplement: PM-145 Rev.2 Maintenance according to Aircraft Maintenance Manual Supplement document: AI/CA-887. | - Limited to model B1900D with installed Rockwell Collins Pro Line 2100 only. The installation of this design change must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 26/02/2016 | Active | 10055033 |
| 1. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 26/02/2016 | Active | 10055039 |
| Master Data List: L2-76AK-MDL-1, Revision 1, dated June 2, 2015. | 1. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 22/08/2016 | Active | 10055038 |
| - Master Data List, AE15136MDL, Issue 0: Aircraft Flight Manual Supplement, AE15136AFMS, Issue 0: Instructions For Continued Airworthiness, AE15136ICA, Issue 1: Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | - As per Aircraft Flight Manual Supplement: EASA. | 08/10/2015 | Active | 10055044 |
| - Master Data List: MDL: 22001134_F: Summary Report: 21001390_C: Flight Manual Supplement: FMA 63001943_C: Maintenance Planning Document Supplement TM 61001092_B: Section 9 Airworthiness Limitations (AIR/ILs) and Certification Maintenance Requirements (CMRs) Only applicable to Serial Numbers 46151 Prior to install... | - As per Flight Manual Supplement: FMA 63001943_C: Section Certificate Limitations- As per Maintenance Planning Document Supplement TM 61001092_B: Section 9 Airworthiness Limitations (AIR/ILs) and Certification Maintenance Requirements (CMRs) Only applicable to Serial Numbers 46151 Prior to install... | 24/03/2017 | Active | 10055045 |
| - Master Data List, AE15136ICA, Issue 1: Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | - As per Flight Manual Supplement: FMA 63001943_C: Section Certificate Limitations- As per Maintenance Planning Document Supplement TM 61001092_B: Section 9 Airworthiness Limitations (AIR/ILs) and Certification Maintenance Requirements (CMRs) Only applicable to Serial Numbers 46151 Prior to install... | 08/10/2015 | Active | 10055047 |
### List of EASA Supplemental Type Certificates

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<th>EASA STC No.</th>
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<th>System Record</th>
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<tr>
<td>1005548</td>
<td>Sabena Techinics BGC</td>
<td>767-200/300/300F/400ER</td>
<td>TCAS 7.1 Upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Data List: Aerosol document No. 0732-01-A-MDL-R00, Revision 0, dated 6 August 2015; Aircraft Flight Manual Supplement, Aerosol document No. 0732-01-A-AFMS(&lt;&gt;,++) Revision 0, dated 3 July 2015; Instructions for Continued Airworthiness, Aerosol document No. 0732-01-AICA-R00, Rev...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>1005507</td>
<td>B/E Aerospace, Inc.</td>
<td>FAA STC ST04284AT</td>
<td>737-600/700/800/900</td>
<td>Installation of the B/E Aerospace LED Lights in the Main Cabin</td>
<td>The Certification Basis for the original product as amended is the following additional or alternative airworthiness requirements: CS25.441 for affected area. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged...</td>
<td>Master Drawing List 2-153358-002 rev B dated Sept 2, 2015; Weight and Balance change document 2-153358-004 rev A dated Aug 14, 2015; Instructions for Continued Airworthiness 2-153358-004 rev A dated Aug 14, 2015; later revisions of the above listed documents approved by EASA in accordance with EASA ED...</td>
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<td>1005508</td>
<td>Wheelen Engineering Company Inc.</td>
<td>FAA STC SA800EA</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>EASA validation of FAA STC SA800EA: Installation of Wheelen Anti-Collision Position Light System - limited to EASA GA aeroplanes (Approved Model List)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Wheelen Anti-Collision Light Systems Installation and Service Manual Document No. 05131, Rev. D dated 6 March 2015 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/40/CF (or subsequent revisions of this decision) and/or th...</td>
<td>Limited to EASA Approved Model List (attached). Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/10/2015</td>
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<td>10055075</td>
<td>Gulfstream Aerospace Corporation</td>
<td>REV. 1 FAA ST04284AT-D</td>
<td>GVI (G650)</td>
<td>Installation of AirCell ATG-4000 High Speed Internet (AHSI) system. This modification provides a high speed two-way data connection from the airborne aircraft to the internet via a network of terrestrial transceiver sites. The systems consists of an Air-To-Ground (ATG) receiver-transmitter, two antennas...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- Index List, ref: GD1.100000/14, dated 18 April 2015; Aircraft Flight Manual Supplement, ref: GD1.100302/001, dated 23 April 2015; Instruction for Continued Airworthiness, ref: CE813/03/001 Revision D dated 16 March 2015; later revisions of the above listed documents approved by EASA in accordance with...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/02/2017</td>
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<td>10055077</td>
<td>Armstrong Aerospace Inc. dba Armstrong Aerospace</td>
<td>FAA STC ST03488CH</td>
<td>AS20-214 A319-112-115</td>
<td>Installation of a Passenger Cabin P/E Upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- Master Data List: Document No. DC164-0055-00, Revision B, dated May 22-2015; later FAA approved revisions.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/10/2015</td>
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<td>EASA STC No.</td>
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<td>10055080</td>
<td>NORTWEST AEROSPACE TECHNOLOGIES INC.</td>
<td></td>
<td>REV. 2</td>
<td></td>
<td>A340-642</td>
<td>Installation of Economy Class Seats and Power Supply System for Virgin Atlantic Airways.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Summary: Northwest Aerospace Technologies, Inc. document No. N5050-D91, Revision 1 dated 29 November 2015 - Instructions for Continued Airworthiness: Northwest Aerospace Technologies, Inc. document No. N5050-D70, Revision A dated 17 September 2015 for later revisions of the above listed.</td>
<td>Prior to installation of this change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/12/2016</td>
<td>Active</td>
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<td>10055082</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>FAA</td>
<td>ST02910CH-D</td>
<td></td>
<td>BD-700-1A10, BD-700-1A11</td>
<td>Installation of a 3-Place Divan</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CR-1 (E) INJURY CRITERIA FOR SIDE FACING DIVANS (SOFAS). The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>JET AVIATION ST. LOUIS INC. INDEX LIST No. 35-2950035 REV NC DATED 28 MAY 2012.</td>
<td>Jet Aviation St. Louis Inc. Installation limitations and follow on appliability requirements document No. 35-1066x02 Rev A dated 26 May 2012 is required.</td>
<td>12/10/2015</td>
<td>Active</td>
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<td>10055084</td>
<td>ESG</td>
<td>ELEKTRONIKSYSTEM TEM-UND LOGISTIK-GmbH</td>
<td>REV. 2</td>
<td></td>
<td>E0635 T3, E0635 T2+, E0635 T3, E0635 P2+, E0635 P2, EC135 P3, E0635 P3, EC135 T2+, EC135 T2+, EC135 T3, EC135 T2, EC135 T1, E0635 T1, EC135 T2, EC135 T1, E0635 T2, EC135 P0, EC135 P1, EC135 P2, EC135 P2+</td>
<td>Mission Management System Workstation Installation of Installation of a Mission Management System (MMS) Workstation to support surveillance, utility survey and search&lt;&gt;&amp;&lt;)&gt;rescue mission&lt;&gt;&amp;&lt;)&gt;mission.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Rotorcraft Flight Manual Supplement 0022-DO-M2-FMS-00 Issue 02: Master Drawing List 0022-DO-M2-MDL-00 Issue 20 - Means of Compliance Records 00059-DO-M2-MOCR-00 Issue 02 - Instructions for Continued Airworthiness 0022-DO-M2-IAC-00 Issue 120 later revisions of the above listed documents (s) approved.</td>
<td>Prior to installation of this change it must be determined that the interrelationship between this change and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2022</td>
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<td>10055102</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td></td>
<td>45</td>
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<td>45</td>
<td>Northern Avionics Waves: wireless in-flight entertainment system Installation of an entertainment system platform with a Wi-Fi router that enables a wireless LAN onboard the aircraft so which passengers can connect their own personal PEDs</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25amdlt;14 for 25.301, 25.305, 25.561, 25.581, 25.601, 25.603, 25.605, 25.609, 25.6, 25.862, 25.711, 25.863&lt;&gt;&amp;&lt;)&gt;25.869, 25.899, 25.1301, 25.1309, 25.1322, 25.135, 25.135&lt;&gt;&amp;&lt;)&gt;</td>
<td>Aircraft Flight Manual Supplement AFMS-NAV-0491 Rev. 0 - WaveES dated 09/2015 for later revisions of the above listed documents approved by EASA Northern Avionics doc: MDL0491 Rev. 0 - Master Document List, dated 15/10/2015 Northern Avionics doc: ICA-NAV0491 - Instructions for Continued Airworthiness.</td>
<td>Prior to installation of this change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadwise effect upon the airworthiness of the product.</td>
<td>13/10/2015</td>
<td>Active</td>
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<td>10055109</td>
<td>HELI-ONE</td>
<td>TCCA STC</td>
<td>SH08-27</td>
<td>S-76A, S-76C</td>
<td>Installation of Baro Altitude Display Mode Switch in accordance with Avionics Design Services Ltd. Master Drawing List MDL0043</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List MDL0043, Revision BM: Maintenance Manual Supplement HCA37634010-730, Revision 2</td>
<td>Prior to installation of this change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadwise effect upon the airworthiness of the product.</td>
<td>14/10/2015</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10055111</td>
<td>VAN RIEMSDIJK ROTTERDAM B.V.</td>
<td>747-4F</td>
<td>747-400F 747-400 (BCF)</td>
<td>F2E-0581 Pallet (19&quot;x125&quot;)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Declaration of Design and Performance - DDP-150581 Issue 4</td>
<td>Compliance Record - CR-150581 Issue 4</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect...</td>
<td>14/10/2015</td>
<td>Active</td>
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<tr>
<td>10055112</td>
<td>SABENA TECHNICS BGC</td>
<td>ATR 72-212 A ATR 72-211/212 ATR 72-201/202 ATR 72-101/102</td>
<td>Second Mode S installation with ELS/ EHS upgrade – upgrade to TCAS 7.1 on ATR (ACAS II 7.0 upgrade)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Applicable regulations at CS 25 Amendment 4 (refer to associated Certification Plan 0762-01-A-CP-R01, section 5A, 6. for details). The requirements for environmental protection.</td>
<td>Master Data List, ref: 0762-01-A-MLD- R06: Instruction for Continued Airworthiness, ref: 0762-01-A-CA- R01 Aircraft Flight Manual Supplement, ref: 0762-01-A-AMS-R06: Detailed revision of the above listed documents approved by EASA.</td>
<td>As per - Section 9 - Limitations / DECLARATION OF DESIGN AND PERFORMANCE - DDP-150581 Issue 4 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect...</td>
<td>14/10/2015</td>
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<tr>
<td>10055113</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A340-212</td>
<td>Large Monitor Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CR-D-01/10057430 Glass Screens Displays / Monitors. The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Master Data List MLD, ES01-2500 Rev. B dated 13.10.2015 or later revisions of the above listed documents approved by EASA.</td>
<td>The installation of this STC is limited to MSN 009.</td>
<td>14/10/2015</td>
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<tr>
<td>10055115</td>
<td>AERODATA AG</td>
<td>ATR 42-600</td>
<td>EFIS interface for AeroFIS system on an ATR 42-500: The scope of this approval is the operation of the ATR 42-500 with the capability of display on the flight deck EFIS (Electronic Flight Instrument System) of the navigation information issued by a Flight Inspection System (AeroFIS), for the purpose of...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: All affected requirements (see Compliance Summary) at amendment 4 of CS25, as an effect to comply. The requirements for enviro...</td>
<td>Airplane Flight Manual Supplement, Aerodata reference AFM-138. For later revisions of the above listed documents approved by EASA. Compliance Summary, Aerodata reference CS-ATR42-01.01 rev. 0 dated 24-Sep-2015. Master Document List, Aerodata reference MLD-A/C-847 rev 3 – dated 13-Oct-2015 Declaration...</td>
<td>1. This approval is limited to ATR 42-500 aircraft MSN 673.2. The aircraft must have the following modifications already embodied (prerequisite configuration): – EASA STC 10047498 rev. 1 (Integration of a Flight Inspection System into an ATR 42-500) - Aerosonde minor change MCA - ATR42-01.02 Rev. 2 (Provis...</td>
<td>14/10/2015</td>
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<tr>
<td>10055127</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A340-212</td>
<td>Tailwind 550 SAT TV Antenna Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List MLD ES01-4420 Rev. B dated 14-Oct-2015 Airplane Flight Manual Supplement AFMS- ES01-4420 Rev. A dated 29-Sep-2015: Instructions for Continued Airworthiness ICA-ES01-4420 Rev. A dated 01-Oct-2015 or later revisions of the above listed documents approved by EASA.</td>
<td>-Per Airplane Flight Manual Supplement AFMS-ES01-4420 and Instructions for Continued Airworthiness ICA-ES01-4420. The installation of this STC is limited to MSN 009 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previous...</td>
<td>15/10/2015</td>
<td>Active</td>
<td>10055127</td>
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<td>10055130</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Replacement of existing TCAS 7.0 by TCAS 7.1 in accordance with Scandinavian Avionics Design document 2087-492- DOC. at issue 1, dated 09-October 2015.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Flight Manual Supplements, Scandinavia Avionics Design Documents 207492-AM-A at issue 1, dated September 3, 2015, for aircraft equipped with Honeywell TPA-100B TCAS II Processor. 2097492-AM-B at issue 2, dated September 17, 2015, for aircraft equipped with ACSS TCAS 2000 RT960/RT-951 Processor...</td>
<td>None</td>
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<td>STC ref.</td>
<td>Movie</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>STANDARD AERO</td>
<td>FAA STC 010569H-D</td>
<td>EMB-145: EMB-145MP EMB-145EP: EMB-145ER EMB-135ER: EMB-135LR EMB-145</td>
<td>Installation of a Reconfiguration of a Cabin Floorplan</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. There are requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>1. Standard Repair Data List 1023035 Rev (J), dated 4 February 2011or later FAA approved revision 2. Airplane Flight Manual Supplementation document number 1033399 Rev (B), dated 4 November 2015 or later FAA approved revision 3. Airplane Flight Manual Supplementation document number 1056686 Rev (A), dated 7.</td>
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<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/02/2019</td>
<td>Active</td>
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<td>DIAMOND J, Inc.</td>
<td>FAA STC SR0231LA</td>
<td>407</td>
<td>Installation of anyone, or a combination of the eight Diamond J Inc.; propulsion system indicators. See continuation in Appendix for indicator part numbers</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA STC SR0231LA</td>
<td>407</td>
<td>Installation of anyone, or a combination of the eight Diamond J Inc.; propulsion system indicators. See continuation in Appendix for indicator part numbers</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>AEROtec &amp; CONCEPT</td>
<td>FAA STC SR0231LA</td>
<td>407</td>
<td>Installation of anyone, or a combination of the eight Diamond J Inc.; propulsion system indicators. See continuation in Appendix for indicator part numbers</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>AGUSTAWESTLANT PHILADELPHIA-HOP Polar</td>
<td>FAA STC SR0231LA</td>
<td>407</td>
<td>Installation of Emergency Flotation System on Bell 412, 412PH helicopters in accordance with Agusta Emergency Flotation System KOP/N 412-4600.04, Rev. DI</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>MECASER AVIATION GROUP S.p.A.</td>
<td>REV. 2</td>
<td>MBB-BK117 D-3 MBB-BK117 D-2 MBB-BK 117 D-2</td>
<td>VIP MB Style Interiors P/N BCC10000032001. The change consists of passenger cabin interior VIP customization, wall layout, seats, small cabinets, entertainment system and other items as detailed in the configuration description documents. Revision 1 is to add compatibility with the</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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**List of EASA Supplemental Type Certificates**
easa.europa.eu
<table>
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<tr>
<th>STC Holder</th>
<th>Rev. level</th>
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<th>Models</th>
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<tr>
<td>WHELEN AEROSPACE TECHNOLOGIES LLC</td>
<td>REV. 2</td>
<td>FAA-STD-SA0279AT</td>
<td>SEE FAA AML SEE EASA APPROVED MODEL LIST</td>
<td>Installation of HID Landing and Taxi Lights</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Design &amp; EASA Report Number 43, Revision F, dated 15 July 2003 or later revisions of the above listed document(s) approved/ accepted beforehand of EASA in accordance with the Technical Implementation Procedures of the EU/USA Bilateral Agreement.</td>
<td>The STC is valid for aircraft types and models approved by EASA listed on the latest approved FAA Aircraft Model List. Prior to installation of this design change/repair it must be determined that the Technical Implementation Procedures of this STC shall remain applicable to the original product.</td>
<td>26/03/2021</td>
<td>Active</td>
<td>10055183</td>
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<tr>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>REV. 1</td>
<td>FAA-SR01052AT</td>
<td>BELL 412, BELL 412EP BELL 205A-1, BELL 212 A8B12, A8B12EP</td>
<td>Replacement High Sky Landing Gear Crossbars for Bell Helicopter/Textron Models 205A-1, 212, 412, 412EP, and LEONARDO S.P.A. Models AB412, AB412EP</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Instructions for Continued Airworthiness Crossbars AA-01136 Rev. L, dated July 30, 2012 or later revisions of the above listed documents approved by EASA in accordance with EASA Decision 2004/04 CF (or subsequent revisions offer this decision) and/or the Technical Implementation Procedures of EU/USA.</td>
<td>Prior to installation of this design change/repair it must be determined that the Technical Implementation Procedures of this STC shall remain applicable to the original product.</td>
<td>08/03/2017</td>
<td>Active</td>
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<td>FEDERATION FRANCAISE DE VOL A VOILE (FFVV)</td>
<td></td>
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<td>201 A, 201 B, 201 B1</td>
<td>Polyurethane paint on gliders C201 &quot;Marianne&quot; A, B, X,Y,Z</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels remain applicable to this certificate/ approval.</td>
<td>BS FFVV 020215 edition 1 dated 19/10/2015 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>23/10/2015</td>
<td>Active</td>
<td>10055190</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td>ERU 190-200 STD ERU 190-200 LR ERU 190-200 IW (195 ART) ERU 190-100 SR ERU 190-100 LR ERU 190-100 IW ERU 190-100 EC2 (LINEAGE 1000)</td>
<td>Aerosonded Supplemental Type Certificate STC 0699-01: Mechanical and electrical installation for personal tablet computer on E190 family aircraft</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s): Enhanced Airworthiness Programme for Aeronautical Systems - ICA ordewis-E08 &quot;Engine Sustained Imbalance ';' ID applicable paragraphs according to CS...</td>
<td>-Aerconesd Master Data List ref. 0689-01-A-2301 Rev. 01 dated 20 October 2015, and Aerconesd AFM Supplement ref. 0689-01-A-2301 Rev. 01 dated 16 October 2015, and Aerconesd Instruction for Continued Airworthiness (IC) ref.0689-01-A-2211 Rev. 05 dated 20 July 2015 or later revisions of the above.</td>
<td>-JBCS-M-00001AFM-01 Issue 1 Airplane Flight Manual Supplement-JBCS-M-00001ICA-01 Issue 1 ICAW Instruction for Cont. Airworthiness-JBCS-M-00001ICA-01A Issue 1 Instruction for Cont. Airworthiness- The following ICA documents contain Airworthiness Limitations-JBCS-M-00001ICA-08 Issue 1 Iridium antenna in...</td>
<td>The limitations as identified in the AFM Supplement are applicable. The mechanical and electrical installation of this STC shall only be used for portable EFBS. It is the operator/responsibility to apply further the operational assessment of EFBS at its National Aviation Authorities. The portable EFB shall...</td>
<td>21/10/2015</td>
<td>Active</td>
<td>10055191</td>
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<tr>
<td>JET AVIATION AG</td>
<td>REV. 3</td>
<td>A320-214</td>
<td>Transfer of STC 10055193 Rev. 2 to new holder/VP Interior Installation A320-214</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s): Enhanced Airworthiness Programme for Aeronautical Systems - ICA ordewis-E08 &quot;Engine Sustained Imbalance ';' ID applicable paragraphs according to CS...</td>
<td>-JBCS-M-00001AFM-01 Issue 1 Airplane Flight Manual Supplement-JBCS-M-00001ICA-01 Issue 1 ICAW Instruction for Cont. Airworthiness-JBCS-M-00001ICA-01A Issue 1 Instruction for Cont. Airworthiness- The following ICA documents contain Airworthiness Limitations-JBCS-M-00001ICA-08 Issue 1 Iridium antenna in...</td>
<td>Limited to S/N 6285</td>
<td>10/10/2017</td>
<td>Active</td>
<td>10055192</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>REV. 1</td>
<td>A320-231/230-233 A320-231/-230-214/215-216</td>
<td>GEE Broadband System Activation</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Applicable regulations at CS 25 Amendment 4 A &amp; B (refer to associated Certification Plan 0699-01-A-2101-02 and 0695-01-A-2101-F-01-R00, section 4 &amp; 5, (h)(i)(v) for details) The...</td>
<td>-Master Data List, Aerconesd document No. 0695-01-A-2301, Revision 2 dated 18 December 2015, Aircraft Flight Manual Supplement, Aerconesd document No.0695-01-A-2301, Revision 2, dated 17 December 2015, Instructions for Continued Airworthiness, Aerconesd document No.0695-01-A-2111, Revision 3, d...</td>
<td>Prior to installation of this design change this STC must be determined that the Technical Implementation Procedures of this STC shall remain applicable to the original product.</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10055196</td>
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## List of EASA Supplemental Type Certificates

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<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. Level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification Basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10055197</td>
<td>SABENA TECHNICS BGC</td>
<td>REV 2</td>
<td></td>
<td>A320-216, -231, -232, -233, A320-211, -212, -214, -215</td>
<td>Ku-band Antenna Provisions Installation</td>
<td>The Certification Basis for the original product as amended by all applicable technical or alternative airworthiness requirements: CRIB-0301-001 (Vibration/buffeting compliance criteria for large external antenna installation). The requirements for environmental protection and the associated noise levels remain applicable to this certificate/ approval.</td>
<td>Master Data Unit, Aerocore document No. 0736-01-A-2103, Revision 1 dated 18 February 2018.Instructions for Continued Airworthiness, Aerocore document No. 0736-01-A-2211, Revision 1, dated 19 October 2015 for airplanes modified per “configuration 1” installation design (as approved by the int...</td>
<td>None</td>
<td>01/02/2016</td>
<td>Active</td>
<td>10055197</td>
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<tr>
<td>10055203</td>
<td>MONARCH AIRCRAFT ENGINEERING LTD</td>
<td>CAA UK RESPONSIBIL</td>
<td>DTY</td>
<td>A321-231</td>
<td>214 PASSENGER RECONFIGURATION</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated noise levels remain applicable to this certificate/ approval.</td>
<td>MONARCH DESIGN SERVICES MASTER DOCUMENT LIST NO. MON/320/25/MDL/3474/SUE C</td>
<td>None</td>
<td>22/10/2015</td>
<td>Active</td>
<td>10055203</td>
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<tr>
<td>10055204</td>
<td>AVONIK ENTHWICKLUNGS GmbH</td>
<td>CABI G2</td>
<td></td>
<td></td>
<td>Installation of Aspen EFD1000H flight display and Mid-Continent 4300-31 audio indicator, in accordance with Avonik Straubing Entwicklungen GmbH, Modification/Modification file no. ASR-2014-046</td>
<td>Flight Manual Supplement No. ASR-2014-046-FMS-05-00-Instructions dated 20/09/2014 or later revisions of the above listed documents approved by EASA.Master Document List No. ASR-2014-046-MDL-01-00 dated 12/09/2014 for Continued Airworthiness for ASR-2014-046-ICA-11-00 dated 12/09/2014 or later revisions of the above listed documents</td>
<td>1) This installation is limited to VFR operations. 2) The aircraft must be equipped with airspeed, altitude, attitude and vertical speed indicators independent from the EFD1000H display. 3) For night operations the aircraft must be equipped with an electro-mechanical attitude indicator independent from the primary attitude indicator.</td>
<td>22/10/2015</td>
<td>Active</td>
<td>10055204</td>
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<tr>
<td>10055205</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>BN2A-21</td>
<td></td>
<td></td>
<td>Installation of Mission Equipment Int. Ballasts on the Empennage inaccordance with S4A Master Drawing List BN003-RPT02.</td>
<td>Flight Manual Supplement BN003-HDK01, issue 1, dated 23 September 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>Limited to BN2A-21, S/N 523 and 531 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/10/2015</td>
<td>Active</td>
<td>10055205</td>
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<td>10055227</td>
<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBIL</td>
<td>DTY</td>
<td>737-800</td>
<td>Cabin Configuration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated noise levels remain applicable to this certificate/ approval.</td>
<td>Tenencia Modification TEN 788 Boeing 737-800 Cabin Configuration Issue 1 dated 22 October 2015</td>
<td>Restricted to ASN 37750</td>
<td>23/10/2015</td>
<td>Active</td>
<td>10055227</td>
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<tr>
<td>10055228</td>
<td>ICELANDAIR EHF</td>
<td>757-300 757-200</td>
<td></td>
<td>Replace existing Audio Selector Panels with Gables Audio Control Panel/PN G27257-04</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated noise levels remain applicable to this certificate/ approval.</td>
<td>- Change/Repair Approval: Icelandic document No. CRA-757-2350-10, Revision A, dated 21 October 2015 Continued Airworthiness Instructions, Icelandic document No. CAI-757-2350-10, Revision IR, dated 10 July 2014 Aircraft Flight Manual Supplement, Icelandic document No. AFM-757-2350-10, Revision IR...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/10/2015</td>
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<td>10055228</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAAC (EC no.)</td>
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<td>Associated Technical Documentation</td>
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<tr>
<td>10055233</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC STO174W1</td>
<td>MYSTERE-FALCON 900</td>
<td>Validation of FAA STC Upgrade from a Honeywell Traffic Alert and Collision Avoidance System CAS-81 (CAS II), to a Honeywell Traffic Alert and Collision Avoidance System CAS-100 (CAS II version 7.1).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25.1316, CS 25.1317 at Amendment 17 for the areas modified affected by this change. The requirements for environmental protection and the associated certified noise and...</td>
<td>Aircraft Flight Manual Supplement ref. EDOC-000001715 revision A, dated 9 February 2015 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/548CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU USAM...</td>
<td>Applicability is limited to MF900 model. The pre-requisite for this approval is the EASA approval ref. 10055239. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will...</td>
<td>26/10/2015</td>
<td>Active</td>
<td>10055233</td>
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<tr>
<td>10055234</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 1</td>
<td>LBA EMZ 5A 1154</td>
<td>DR 400/180 S DR 400/180 R DR 400/180</td>
<td>GO2015-57 DR 400 Exhaust System Rev. 1 Transfer to ACC Columbia Jet Service GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels. CS-36 Amd. 3.</td>
<td>According to Gomolib Master Document List No. MDOLL 2037202 rev0 20150015STC to be operated in accordance with Gomolib AFM-S Document No. SPOH_2037202_DRA 0_180r rev0 20150015 and SPOH_2037202_DRA 0_180f rev0 201509 15r later revisions of the above listed documents approved by EASA to be installed later...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/01/2022</td>
<td>Active</td>
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<td>10055235</td>
<td>L’ATELIER DE LA MONTGOLFIERE</td>
<td>FIREFLY 8</td>
<td>STC ADM 01-2015Schneider fire balloon bottom and installation under Firefly 8 envelope</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplement manual de vol F-GZAP selon STC ADM 01-2015 for later revisions of the above listed documents approved by EASA.</td>
<td>Modification limited to Balloon Firefly 8 an Fl-BB 2022</td>
<td>26/10/2015</td>
<td>Active</td>
<td>10055235</td>
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<td>10055239</td>
<td>DASSAULT FALCON JET</td>
<td>FAA STC SA3965SW-D</td>
<td>MYSTERE-FALCON 900</td>
<td>Installation of Bandix CAS-81 TCAS System on MF900</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement, Report No. 36255-1, Rev. C. dated 21 August 1995. Airplane Flight Manual Supplement, Report No. 36255-2, Rev. NC dated 21 March 2000. Airplane Flight Manual Supplement, Report No. 36255-3, Rev. NC dated 07 September 2000 or later revisions of the above listed documents applicable...</td>
<td>Applicability is limited to MF900 model The EASA approval ref. 10055239 cannot be installed as a final configuration, but only as a pre-requisite for the EASA approval ref. 10055233. Prior to installation of this design change it must be determined that the interrelationship between this design change and...</td>
<td>26/10/2015</td>
<td>Active</td>
<td>10055239</td>
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<tr>
<td>10055241</td>
<td>I.S.E.I INTERNATIONAL SERVICES ELECTRONIQUE INFORMATIQUE</td>
<td>172S 172N</td>
<td>Installation of SAFETYPLANE VS System</td>
<td>CRI F-01 issue 03 Lithium Battery InstallationsThe requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>List of Documents Project 0010000505-001 N10-Cessna 172R-S-SAFETYPLANE-VS Edit A Revision 02 dated 7 October 2015N1-Cessna 172R-S-9-SPM-AAEROPLANE FLIGHT MANUAL SUPPLEMENT- E-A-01 Rev.0 dated 06 August 2015 for later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>26/10/2015</td>
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<td>10055241</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SA09122AC-D</td>
<td>185E, A185F, A185F 185, 185A, 185B, 185C, 185D</td>
<td>Installation of 3-blade automatic flight guidance system, Model ST-650</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. NEW-PAGE The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>Bulletin No. 750 Revision 4, dated 4-25-01 and Master Drawing List No. 92963 Revision D, dated 4-25-01 (14 Volt System).</td>
<td>1. Also eligible on the above models (landplane/floatplane/amphibian) when equipped with STC SA1441WE (Robertson STOL) or STC SA09122AC (Horton STOL) and/or STC SA2529WE (inertial log) 2. FAA/AAT approved Supplemental Flight Manual, PN 903114, dated 11-06-98 is required for Cessna Models 185</td>
<td>12/10/2015</td>
<td>Active</td>
<td>10055247</td>
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<tr>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>SR20</td>
<td>Installation of 3-blade propeller MTV-12-D/188-596 on Cirrus SR20</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. This certification/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels: CS-36 Amdt. 3 and FAR 36 Amdt. 30</td>
<td>According to MT-Propeller Master Document List No. E-2630 for later revisions of the above listed documents approved by EASA. To be installed in accordance with MT-Propeller AFM-S Document No. E-2630 for later revisions of the above listed documents approved by EASA. To be installed in accordance with MT-Propeller AFM-S Document No. E-2630 for later revisions of the above listed documents approved by EASA To be installed in accordance with MT-Propeller AFM-S Document No. E-2630 for later revisions of the above listed documents approved by EASA. To be maintained i.a.w.</td>
<td>09/11/2015</td>
<td>Active</td>
<td>10055250</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>DR-400/99 DR-400/200 R DR 400/180 R DR 400/140 B DR 400/120 D</td>
<td>Installation of 3-blade constant speed propeller MTV-6-A/190-69 on C.P.R. DR400 series modified with Tefen 250-300 engine per STC10014219</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. This certification/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels: CS-36 Amdt. 3 and FAR 36 Amdt. 30</td>
<td>According to MT-Propeller Master Document List No. E-2711 for later revisions of the above listed documents approved by EASA. To be installed in accordance with MT-Propeller AFM-S Document No. E-2711 for later revisions of the above listed documents approved by EASA. To be installed in accordance with MT-Propeller AFM-S Document No. E-2711 for later revisions of the above listed documents approved by EASA. To be installed in accordance with MT-Propeller AFM-S Document No. E-2711 for later revisions of the above listed documents approved by EASA. To be maintained i.a.w.</td>
<td>26/10/2015</td>
<td>Active</td>
<td>10055251</td>
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<tr>
<td>AVIGATION DEFENSE SERVICE</td>
<td>SR22</td>
<td>AVD-MSR22-95-ISR4A Issu01 dated 12/10/2015 - Activation of ISR mission system NEW-PAGE</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this/ certification/approval.</td>
<td>Instructions for continued Airworthiness ICA-ADV- SR22-ISR Issue 01 dated 14 September 2015/ Master Document List MDL-ADV-ISR4a dated 12 October 2015/ POH and EASA approved AFM supplement SAFM-ADV- SR22-ISR Revision 01 dated 12 October 2015 for later revisions of the above listed documents approved by EASA. To be operated in accordance with MT-Propeller AFM-S Document No. E-2708 for later revisions of the above listed documents approved by EASA. To be maintained i.a.w.</td>
<td>27/10/2015</td>
<td>Active</td>
<td>10055257</td>
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<tr>
<td>AVIGATION DEFENSE SERVICE</td>
<td>REV. 2</td>
<td>AVD-MSR22-95-ISR3A Issue 01 dated 12/10/2015 - Carriage of ISR mission external loads</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>Instruction for continued Airworthiness ICA-ADV- SR22-ISR Issue 01 dated 14 September 2015/ Master Document List MDL-ADV-ISR3A dated 12 October 2015/ POH and EASA approved AFM supplement SAFM-ADV- SR22-ISR Revision 01 dated 12 October 2015 for later revisions of the above listed documents approved by EASA. To be operated in accordance with MT-Propeller AFM-S Document No. E-2607 for later revisions of the above listed documents approved by EASA. To be maintained i.a.w.</td>
<td>10/11/2015</td>
<td>Active</td>
<td>10055258</td>
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List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<th>System Record</th>
</tr>
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<tbody>
<tr>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>LBA SAT263</td>
<td>F260, F260B, F260C, F260D</td>
<td>Installation of 3-blade constant speed propeller MTV-B-B/185-50 onAlenia Aermacchi F260 series</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection and a change to the certified noise or emissions level of the original product.</td>
<td>According to MT-Propeller Master Document List No. E-2732 (Tide operated in accordance with MT-Propeller AFM-MS Document No. E-7760) later revisions of the above listed documents approved by EASA to be installed iaw. Installation Instructions/Dow. No. E-783, or later/Revision To be maintained a/w MT-...</td>
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<td>27/10/2015</td>
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<td>10055270</td>
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<tr>
<td>HONEYWELL INTERNATIONAL, Inc.</td>
<td>REV. 1</td>
<td>FAA STC SR02461SE</td>
<td>A9139</td>
<td>Installation of a Honeywell ASPIRE 200 Intermediate Kit (6A) SATCOM System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Drawing List (MDL) MOD- A9139ASPIRE200-5201, Revision C, dated 13 December 2016, or later FAA-approved revision. 2. Instructions for Continued Airworthiness (ICA) documents in &lt;...&gt; or &lt;...&gt; FAA-approved as applicable ICA MOD-A9139ASPIRE200-7601, Revision E, dated 9 November 2016 (CA MO...</td>
<td></td>
<td>06/07/2017</td>
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<tr>
<td>NORTHERN AERONAUTICS S.R.L.</td>
<td></td>
<td></td>
<td>45 (LEARJET 45)</td>
<td>Aeroacoustics 100 (In-Flight Internet/Satcom)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amdt. 14The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable.</td>
<td>CCL No. CCL0500-001 Rev. of the latest revision of the above listed documents approved by EASA.</td>
<td></td>
<td>28/10/2015</td>
<td>Active</td>
<td>10055274</td>
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<tr>
<td>SABENA TECHNICS BOD</td>
<td>REV. 4</td>
<td>A330-223</td>
<td>A330-223 EpicSafe Protection System (SPS) modification.This modification has been achieved in three phases. The initial phase has prepared the aircraft belly area to receive a possibly relocating antennas, drain mast, and provisioning some other equipment. The second phase has checked the functionality. The third phase has prepared the aircraft belly area to receive a possibly relocating antennas, drain mast, and provisioning some other equipment. The second phase has checked the functionality.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equipment Safety Finding(s): B-401-003: Vibration &amp; Buffeting/Complex criteria for large external antenna installation. The requirements for environmental protection and...</td>
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<td>NATIONAL AIRCRAFT SERVICES, INC.</td>
<td>FAA STC ST0439CH</td>
<td>727-200, 727-200F</td>
<td>Install National Aircraft Service, Inc. Vent Door in a main cabin window of 727 airplanes, modified with a Cargo Door/ Trim</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. FAA STC ST0439CH/N/E/NEW/PAGE requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>National Aircraft Service Inc. Drawings List RXV-DL 14, Rev. A, dated March 22, 2001/Installation Manual RXV-727-14, No Rev, dated March 4, 2001 or later/FAA-Approved revision.</td>
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<td>10055297</td>
<td>KEVEL</td>
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<td>FAA STC SA2006CH</td>
<td></td>
<td>Installation of air conditioning system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Kelly Aerospace Installation Instructions as listed on AML No. SA2006CH.</td>
<td>1. Airplane Flight Manual Supplement as listed on the FAA Approved Model List (AML) No. SA2006CH. is required on board the modified aircraft. 2. Instructions for Continued Amortiness (ICA) as listed on FAAApproved Model List (AML) No. SA2006CH. should be incorporated into the aircraft Maintenance...</td>
<td>18/12/2015</td>
<td>Active</td>
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<tr>
<td>10055296</td>
<td>GYRO-STABILIZED SYSTEMS, LLC</td>
<td>REV. 2</td>
<td>FAA STC SR20447AK</td>
<td>R66</td>
<td>Installation of ZW Aviation LLC Utility Camera Mount in accordance with Master Drawing List (MDL) R66M-010</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement for the installation of R66M-010 Utility Camera Mount, document number R66M-007 revision P, dated May 13, 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce a new adverse effect upon the airworthiness of the product.</td>
<td>25/10/2022</td>
<td>Active</td>
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<tr>
<td>10055312</td>
<td>HELI-ONE</td>
<td></td>
<td>TCAP STC SH01-17</td>
<td>S-76A, S-76C</td>
<td>Installation of Water Activated Rotation System in accordance with Heli-One Master Drawing List CHS76-32-001 MDL R66M-007</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement CNS CHS76-32-001, Revision A or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement Master Drawing List CHS76-32-001 MDL Revision BMaintenance Manual Supplement MS8 CHS76-32-001</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce a new adverse effect upon the airworthiness of the product.</td>
<td>30/10/2015</td>
<td>Active</td>
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<tr>
<td>10055322</td>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 1</td>
<td>HAWKER 1000</td>
<td></td>
<td>AC55 TCAS Change 7.1 Upgrade with P/N 4066010-914 TCAS 2000 - RT-951 Change 7.1 Upgrade with P/N 7517900-55200.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>328 Design GmbH AFM supplement AFMS022B344Z0002 Issue A released 29 Oct. 2015 for TCAS/AFM supplement AFMS022B344Z0002 Issue A released 18 Sept. 2015 for TCAS2000 Change 7.1 or later revisions of the above listed documents approved by EASA Master Data List MD01/2010 Issue C from 30 Oct. 2015</td>
<td>The approval is limited to TCAS II Change 7.1 Upgrade with computer P/N4066010-914 and TCAS 2000 RT-951 Change 7.1 Upgrade with computer P/N 7517900-55200. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a new adverse effect upon the airworthiness of the product.</td>
<td>27/09/2018</td>
<td>Active</td>
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<tr>
<td>10055330</td>
<td>BOND HELICOPTERS EUROPE Ltd.</td>
<td></td>
<td>CAA UK RESPONSIBIL</td>
<td>EC135 T3(CPDS)</td>
<td>NVIS for HIOW Air Ambulance</td>
<td>Flight Manual Supplement for HOW FMS15-002 - Issue 2 dated 21 October 2015 or later revisions of the above listed documents approved by EASA - Design Report DR/ 115-501- Issue 1, dated 16 September 2015 - Instructions for Continued Amortiness BOND/MD0/ 115, HIOW AirAmbulance, NVIS, EC135 - ...</td>
<td>Any subsequent deviation from the cockpit or cabin configuration specified in the STC Type Design may affect the compatibility of the NVIS and will require an engineering evaluation to assess the effect of any change on NVG compatibility. The engineering evaluation must be approved by the aircraft manufacturer.</td>
<td>02/11/2015</td>
<td>Active</td>
<td>10055330</td>
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<td>STC Holder</td>
<td>STC No.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>AIR ALLIANCE GmbH</td>
<td>10055335</td>
<td>STC Holder</td>
<td>LEARJET 550/555/650</td>
<td>LEARJET 310/3A/35/35A</td>
<td>Operation of TCAS II Change 7:1Software Upgrade from TCAS II Version 7.0 to 7.1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirement for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Air Alliance Aircraft Flight Manual Supplement, doc. No. 2015_014_LJXJ_0506, rev. 01, dated 27 October 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>02/11/2015</td>
<td>Active</td>
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<td>RHEINLAND AIR SERVICE GmbH</td>
<td>10055381</td>
<td>REV. 1</td>
<td>H30, J30A, J30GT, J30GTF</td>
<td>Installation of Dual Garmin G800 System.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CS-23, Amendment 323, 29, 301, 303, 305, 307, 561,601, 603, 605, 609, 611, 613, 771, 773,777, 867, 1301, 1303, 1305, 1309, 1311, 1321, 1322, 1323, 1325, 1326,1327, 1329,1..</td>
<td>Flight Manual Supplement G800 RAS-14-01-02- FMS-00 G800, Rev. 0Flight Manual Supplement ESI-2000 RAS-14-01-02- FMS-00 ESI-2000, Rev. 0Flight Manual Supplement PSS800 RAS-14-01-02- FMS-00 PSS-035D, Rev. 0. Installation of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise/de in the airworthiness of the product.</td>
<td>26/02/2016</td>
<td>Active</td>
<td>10055381</td>
</tr>
<tr>
<td>ONSBOARD SYSTEMS INTERNATIONAL</td>
<td>10055382</td>
<td>FAA SR0244SE</td>
<td>R 66</td>
<td>Cargo Hook Kit</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Rotorcraft Flight Manual Supplement Onboard Systems Cargo Hook/ Suspension System for the Robinson R66 R66-121-058-00, Rev. 0, dated 20 July 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise/de in the airworthiness of the product.</td>
<td>05/11/2015</td>
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<td>10055382</td>
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<td>AERODATA AG</td>
<td>10055385</td>
<td>B300</td>
<td>Installation of AD-APIS-0130 in B300</td>
<td>Installation, Operation and Maintenance according to the Master Drawing, list number MDL-A/ C-872 Rev. 2 dated 2nd November 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>Installation, Operation and Maintenance according to the Master Drawing, list number MDL-A/C-872 Rev. 2 dated 2nd November 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise/de in the airworthiness of the product.</td>
<td>05/11/2015</td>
<td>Active</td>
<td>10055385</td>
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<tr>
<td>HELI-ONE</td>
<td>10055386</td>
<td>TCFA STC SH37.G0</td>
<td>S-76C</td>
<td>Installation of Flight Data Monitoring System in accordance with Heli-One Approval Control Document CHS76-31-013 AC.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Approval Control Document CHS76-31-013 ACD, Revision 0Maintenance Manual Supplement MM8-013, Revision A</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise/de in the airworthiness of the product.</td>
<td>05/11/2015</td>
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<td>STC Holder</td>
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<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>FAA ST02648CH-D</td>
<td>Installation of a Wireless Local Area Network (WLAN)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List 35-409003, Revision D, dated January 29, 2009. The FAAApproved AFMS document defined by the Master Data List is required.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>03/07/2024</td>
<td>Active</td>
<td>10053056</td>
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<tr>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 1</td>
<td>A340-313</td>
<td>Installation of Thales Avionics TopSeries A340-300 Cabin reconfiguration/Replacement of Business, Premium Economy and Economy Class seats, galleys, stowages, IFE and ISPS. Replacement of mid cabin lavatories rework of cabin crew rest module Rev 1. Transfer of STC from SR Technics Switzerland to GROWING Engineering Partners GmbH</td>
<td>Aircraft Flight Manual Supplement S:OAED. 000301-FAE, rev 00Master Document List S:OAED. 000301-MDL, rev 00Instructions for Continued Airworthiness S:OAED.000301-ICA, rev 00</td>
<td>CAGT Attendant Information Panel (API) smoke detection alerts for testltravellers are incorrectly identified within the CIDS system, namely lavatory L35 appears as L37 and L36 appears as L38. OHVD Loudspeakers (the Economy Class do not function properly during video announcement with the design outlook).</td>
<td>20/04/2021</td>
<td>Active</td>
<td>10054301</td>
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<tr>
<td>KNSI AVIATION ENGINEERING LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>757-200</td>
<td>Replacement of Business Class and Economy Class Seats</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>KNSI Master Document List No.146055-MDL-001-S.R Revision 5 dated 01October 2015</td>
<td>Restricted to ASN 23453 and 26246</td>
<td>09/11/2015</td>
<td>Active</td>
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<td>THALES AVIONICS Inc.</td>
<td>FAA STC ST01151LA</td>
<td>A340-642</td>
<td>Installation of Thales Avionics TopSeries i400 Infight Entertainsystem</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Thales Avionics Index List, Drawing No. 25878, Revision A, dated June 18, 2010 for Continued Airworthiness (ICA), Thales Avionics Inc. Document No. 25888, Rev. N, dated March 15&gt; (&lt;,&gt; 2010</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product. This modification may not be installed after Aug 2010.</td>
<td>13/11/2015</td>
<td>Active</td>
<td>10054440</td>
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<td>AVIATION COMMUNICATION ANDSURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>FAA STC ST02530LA</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Installation of T2CASI with TCASI version 7.1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List ACSS document 890044-001 rev. No. 7 dated October 2010 Including all revisions to that date.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>09/11/2015</td>
<td>Active</td>
<td>10054443</td>
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<tr>
<td>ONTIC ENGINEERING ANDMANUFACTURING, Inc.</td>
<td>FAA STC SA4868M</td>
<td>727-200</td>
<td>Installation of a 2300-07-01 or 2307-02-1 digital FuelQuantity gauge System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Smiths Industries Master Drawing 0016-ED rev. V, dated July 7, 1999 Smiths Industries Airplane Flight Manual Supplement No. 001-ED Revision F, dated September 13, 1999 for revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/48/EC (or subsequent revisions).</td>
<td>Aircraft which incorporate a 2-cell centre tank are not approved under this STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>10/11/2015</td>
<td>Active</td>
<td>10054445</td>
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<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>FAA STC ST02543A</td>
<td>767-200-300-400ER SERIES</td>
<td>Installation of Dart Aerospace Evacuation Slide/Raft Modification Kit Part Number 603.8403.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirement for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Dart Aerospace Master Document List MD07/7-1, Revision F, dated August 12, 2015</td>
<td>Each Dart Aerospace Evacuation Slide/Raft Modification Kit Part Number 603.8403 is installation limited to the Goodrich FWD/ AFT evacuation slide/kit Part Number 101651-303. The instructions for Continued Airworthiness (CA) Dart Aerospace Document No. ICA76-1, Revision NIC, dated January 27, 2015 or later.</td>
<td>11/11/2015</td>
<td>Active</td>
<td>1005447</td>
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<tr>
<td>MARSHALL AVIATION SERVICES LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BAE 146 SERIES 300</td>
<td>BAE 146 SERIES 200</td>
<td>BAE 146 SERIES 100</td>
<td>To introduce Universal UNS-1/F Flight Management System as an Upgrade to the Existing UNS-1L. Currently installed on BAE 146 Series Aircraft Under EASA A.5.02889.</td>
<td>The Certification Basis for the original product as amended by this modification. Additional or alternative airworthiness requirements: JAR 25.1322, JAR 25.1431, JAR 25.1529. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>11/11/2015</td>
<td>Active</td>
<td>1005463</td>
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<tr>
<td>AIM AVIATION (JECCO) LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>747-8</td>
<td>WHNCo – 747-8 Dog Box</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AIM AVIATION (JECCO) LTD MODIFICATION NO. MAREB1-13 AND COMPLIANCE CHECKLIST NO. 11894 ISSUE DATED 02 NOVEMBER 2015</td>
<td>Each Dart Aerospace Evacuation Slide/Raft Modification Kit Part Number 603.8403 is installation limited to the Goodrich FWD/ AFT evacuation slide/kit Part Number 101651-303. The instructions for Continued Airworthiness (CA) Dart Aerospace Document No. ICA76-1, Revision NIC, dated January 27, 2015 or later.</td>
<td>11/11/2015</td>
<td>Active</td>
<td>1005465</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>MBB-BK 117 C-2</td>
<td>Provisions for surveillance operations / Installation of surveillance equipment in order to enable the later installation of a separate airdata equipment. Further details are shown in the project description Doc. No.: 084-PD-STC-AT</td>
<td>Provisions for surveillance operations / Installation of surveillance equipment in order to enable the later installation of a separate airdata equipment. Further details are shown in the project description Doc. No.: 084-PD-STC-AT</td>
<td>The Certification Basis for the original product as amended by this modification. Additional or alternative airworthiness requirements: The following paragraphs of CS-29 AMDT 3 are applicable to this change: 29.1, 29.21, 29.25, 29.27, 29.29, 29.45, 29.49, 29.51, 29.64, 29.65, 29.67, 29.75, 29.79, 29.141, 29…</td>
<td>AFM-STC-AT-084; Flight Manual Supplement, Rev. No. 0, dated 03/07/2014; AMDT 3 of CS-29 AMDT 3</td>
<td>Limited to Burt Rutan Aircraft Serial Number 9960. Prior to installation of this design change it must be determined that the intermediate relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/11/2015</td>
<td>Active</td>
<td>1005472</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>KH-950 C</td>
<td>KH-950 antenna provision installation.</td>
<td>KH-950 antenna provision installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation and Maintenance according to the Master Data &amp; Document AE15184MDL Rev. 0, dated 11/11/2015 to 102/11/2015, or later revisions of the above listed documents approved by EASA.</td>
<td>Applicability limited to all MSN Ubs and UC only. Prior to installation of this design change it must be determined that the intermediate relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/11/2015</td>
<td>Active</td>
<td>1005475</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>737-700</td>
<td>Communication - Activation of CPOOLC</td>
<td>Communication - Activation of CPOOLC</td>
<td>The Certification Basis for the original product as amended by this modification. Additional or alternative airworthiness requirements: Special Condition (s): CHI-F-01 “Data Link Services for the Single European Sky” Issue 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>, Supplemental Type Definition Document, Ref. STDD23-13/22M, Issue 1, dated August 25, 2015; Airplane Flight Manual Supplement, Rev. A-00-13/52-AFM, Iss.Rev. 02/00, dated July 09, 2015; Instructions for Continued Airworthiness, Rev. A-00-13/52-10A, Iss. Rev. 01/00, dated Sep 20, 2013 or later.</td>
<td>As per Airplane Flight Manual Supplement, Ref. A-00-13/52- AFM, Iss.Rev.02/00, dated July 09, 2015. Prior to installation of this design change it must be determined that the intermediate relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/11/2015</td>
<td>Active</td>
<td>1005479</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>JET AVIATION AG</td>
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<td></td>
<td>B747-400</td>
<td>B747-400 Main Deck Reconfiguration</td>
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<td></td>
<td></td>
<td>11/13/2015</td>
<td>Active</td>
<td>10055493</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td></td>
<td></td>
<td>MOD 15180</td>
<td>MOD 15180 - TCAS Change 7.1 with Honeywell suite</td>
<td>Master Data List, Ref. JBSL-M-0037ICA-02, Issue 1., - Master Data List, Ref. JBSL-M-0037ICA-01, Issue 1.,</td>
<td></td>
<td></td>
<td>11/13/2015</td>
<td>Active</td>
<td>10055494</td>
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<tr>
<td>DASSAULT FALCON JET</td>
<td></td>
<td></td>
<td>FAA SB233SW-D</td>
<td>FALCON 2000</td>
<td>Installation of the Basic Oxygen System.</td>
<td></td>
<td></td>
<td>11/13/2015</td>
<td>Active</td>
<td>10055495</td>
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<tr>
<td>DASSAULT FALCON JET</td>
<td></td>
<td></td>
<td>FAA SB233SW-D</td>
<td>FALCON 2000</td>
<td>Avionics Rack Installations.</td>
<td></td>
<td></td>
<td>16/11/2015</td>
<td>Active</td>
<td>10055502</td>
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<tr>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>DASSAULT FALCON JET</td>
<td>Installation of the Electrical Feedthru – Pressure Vessel.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Drawing List Report FZM/0120, Rev 05 dated 05 April 1995.</td>
<td>Applicability is limited to F2800 S/N 025. Prior to installation of this design change it must be determined if the interrelationship between this design change and all other previously installed design change and/or repair will introduce噪音 adverse effect upon the airworthiness of the product.</td>
<td>16/11/2015</td>
<td>Active</td>
<td>10055503</td>
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<tr>
<td>DASSAULT FALCON JET</td>
<td>Printed Circuit Board Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Drawing List Report FZM/0120, Rev 05 dated 21 June 1995.</td>
<td>Applicability is limited to F2800 S/N 025. Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise adverse effect upon the airworthiness of the product.</td>
<td>16/11/2015</td>
<td>Active</td>
<td>10055506</td>
<td></td>
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<tr>
<td>ETHAD AIRWAYS</td>
<td>LOPA Change from 20j/120Y to 24j/24P/70Y</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td></td>
<td></td>
<td>16/11/2015</td>
<td>Active</td>
<td>10055508</td>
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<td>ENGINEERING L.L.C.</td>
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<td>16/11/2015</td>
<td>Active</td>
<td>10055506</td>
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<tr>
<td>DASSAULT FALCON JET</td>
<td>Installation of the Air Distribution System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Drawing List Report FZM/0120, Rev 05 dated 20 May 1996.</td>
<td></td>
<td>08/12/2015</td>
<td>Active</td>
<td>10055511</td>
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<tr>
<td>TURK HAVA YOLLARI</td>
<td>Wireless IFE system installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative requirements:</td>
<td>AFM Supplement CE-4421001-FF01 rev 0 dated 12.10.2015 or later, revisions of the above listed documents approved by EASA.AMM Supplement CE-4421001-FF01 rev 0 dated 11.05.2015 CE-4421001-AM01 rev 01 dated 15.11.2015</td>
<td>Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise adverse effect upon the airworthiness of the product.</td>
<td>17/11/2015</td>
<td>Active</td>
<td>10055513</td>
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<td>YOLLARI TECHNIK A.S.</td>
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<td>17/11/2015</td>
<td>Active</td>
<td>10055515</td>
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<td>(TURKISH TECHNIC Inc.)</td>
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<td></td>
<td>17/11/2015</td>
<td>Active</td>
<td>10055515</td>
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<tr>
<td>NORTHERN AVIONICS</td>
<td>Secondary Cycle counter for Breeze Eastern host</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Chap. 4 ALS included in Rev. 09/10/2015 or earlier revisions of the above listed documents approved by EASA. Compliance Checklist CCL0915, Rev 0 dated 9.10.2015 Master Document List MDL0915, Rev 0 dated 9.10.2015 Drawing 25-NAV0915-00 Rev 0 dated 9.10.2015</td>
<td>The installation of this Secondary Cycle Counter is limited to the following aircraft:</td>
<td>17/11/2015</td>
<td>Active</td>
<td>10055515</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>STC Holder</td>
<td>Implementation of ES ADS-B OUT capability with transponder TDR-94D (Mod.AHSE-66400).</td>
<td>The Certification Basis (CB) for the original product contains applicable; this certificate/approval. The requirements for environmental/protect and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to the certificate/approval.</td>
<td>- Standard Aero Master Data List 1026666, Revision (Q), dated October17, 2014 . - Instructions for Continued Airworthiness Document 1026668, revision(D) dated July 23, 2014 in addition to the basic airplane maintenance manuals.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>18/11/2015</td>
<td>Active</td>
<td>10055532</td>
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<tr>
<td>STANDARD AERO AVIATION SERVICE ES, L.L.C.</td>
<td>Installation of Ku Band Satellite System Provisions.</td>
<td>The Certification Basis (CB) for the original product contains applicable; this certificate/approval. The requirements for environmental/protect and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to the certificate/approval.</td>
<td>Standard Aero Master Data List 1024000, Revision (F), dated October25,2014.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>18/11/2015</td>
<td>Active</td>
<td>10055533</td>
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<tr>
<td>HELI-ONE (NORWAY) AS</td>
<td>Implementation of ES ADS-B OUT capability with transponder TDR-94D (Mod.AHSE-66400).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. Deviation from CS-ACNS.D.ADSB.105 as per CRI F-01, issue 3. The following paragraph(s) of a later amendment CS 29 Amend 3 for the following requirements: 29.1, 29.29, 29...</td>
<td>AS 332/L1 Flight Manual Supplement H1N.239 &quot;1590 ES ADSB-OUT&quot; Initial Issue, 16.11.2015 for later revisions of the above listed documents approved by EASA-Product Change Declaration AHSE-66400, Initial Issue. 30.10.2015 Master Drawing AHSE-64407, Rev. A. 10.11.2015 Certification Program AHSE-64403 Rev. C.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>24/11/2015</td>
<td>Active</td>
<td>10055540</td>
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<td>AEROSERVIS, S.R.O.</td>
<td>Introduction of SW version 7.1 to aircraft currently equipped with Honeywell CAS-67A TCAS II as per document AE/162-001P</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) of a later amendment FAR 23.5, Amendment, 1309, 49, 1431, 49, 1581, 50, The requirements for environmental protection and the associated certif...</td>
<td>- STC AE/162 Master Document List, Doc. No. AE/162000X, rev D(&lt;&gt;)&lt;&gt;, dated 06.11.2015, L-410 UVP-E Flight Manual Supplement Doc. No. STC AE/162-DLP1&lt;(&gt;,&lt;&gt;TCAS II Honeywell CAS-67B (SW ver. 7.1), initial issue, dated September10/15, L-410 UVP-E Flight Manual Supplement Doc. No. STC AE/162-DLP2, TCASII.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>19/11/2015</td>
<td>Active</td>
<td>10055546</td>
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<td>3.14159265358979323846...</td>
<td>Installation, Provisions for Passenger Life Vest</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) of a later amendment FAR 23.5, Amendment, 1309, 49, 1431, 49, 1581, 50, The requirements for environmental protection and the associated certif...</td>
<td>Installation in accordance with B/E Aerospace Modification Data Summary JS54007 Rev NC dated 11 May 2015 and later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canada Bilateral Agreement JS44041 Rev A EASA Cert Basis Compliant...</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>19/11/2015</td>
<td>Active</td>
<td>10055550</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>Introduction of TCAS Change 7.1</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) of a later amendment JAR 25 Change 10-25.1301; 25.1306; 25.1431; 25.1581; 25.1585, The requirements for environmental protection and the associ...</td>
<td>STC Twenty One Ltd Flight Manual Supplement S21.EMS-0124 Issue 2 or later revisions of the above listed documents approved by EASA STC Twenty One Ltd Master Document List S21.MDL-1274 Issue 3.</td>
<td>None</td>
<td>20/11/2015</td>
<td>Active</td>
<td>10055557</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>1005570</td>
<td>L2 CONSULTING SERVICES Inc.</td>
<td>FAA ST110375C</td>
<td>777-300, 777-300ER, 777-200, 777-200LR</td>
<td>Installation of SAT-2100B satellite communication system and interface to aircraft systems</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Data List L2 Consulting Services document No. 2015-B61-MDL-01, Revision N, dated 11 August 2015.</td>
<td>None</td>
<td>20/11/2015</td>
<td>Active</td>
<td>1005570</td>
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<tr>
<td>1005575</td>
<td>SABENA TECHNICS BOC</td>
<td>REV. 1</td>
<td>737-900, 737-900ER, 737-700, 737-800</td>
<td>TCAS 7.1 upgrade on B737-The modification consists in the upgrade of the existing TCAS 7.0 systems to TCAS 7.1 system, only computer unit is modifiedRev 1. In addition to the initial release of this STC, Revision 1 includes the associated Software update for Rockwell- Collins (Post Mod PN32-1269-035). Rev:</td>
<td>Certification Basis for the original product as amended by the following additional or alternative environmental requirements: The following paragraph(s) are applicable to this STC.</td>
<td>Master Data List (MDL), ref: 0685-01-A-MDL-R01, revision 4.</td>
<td>None</td>
<td>26/05/2016</td>
<td>Active</td>
<td>1005575</td>
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<td>1005588</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>REV. 1</td>
<td>ATR 72-212 A, ATR 72-211, ATR 72-212 ATR 72-201, ATR 72-202 ATR 72-101, ATR 72-102</td>
<td>ACAS II VERSION 7.1 UPGRADE</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>CS-25 Am01, 17; JAR 21003, CPL0101 Ed. 6. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>72019-LHD01 Flight manual supplement Ed4 or later revision of the above listed documents approved by EASA 72019-SB001 Service Bulletin E372019-HDK02 Maintenance Manual Supplement Ed1.</td>
<td>None</td>
<td>23/11/2015</td>
<td>Active</td>
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<td>1005592</td>
<td>STIRLING DYNAMICS GMBH</td>
<td>REV. 4</td>
<td>CL-600-2C10 (RJ 700, 701, 702) CL-600-2C10</td>
<td>LOPA Change Rev. 4: Transfer of STC</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>V-Planes GMBH &amp; Co.KG Master Document List No. 215096-B-2610-001 Issue 3 dated 30 October 2015: later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the installation of this change/repair to this product will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2022</td>
<td>Active</td>
<td>1005592</td>
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<tr>
<td>1005596</td>
<td>DASSAULT FALCON JET</td>
<td>FAA SAB464SW-D</td>
<td>FALCON 2000</td>
<td>Installation of the Miscellaneous Structures (General/Mechanical)</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Drawing List Report F2M-0707, Rev.D dated 19 June 1997</td>
<td>Applicability is limited to F2000 S/N 025. Prior to installation of this design change, it must be determined that the installation of this design change will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/11/2015</td>
<td>Active</td>
<td>1005596</td>
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<tr>
<td>1005604</td>
<td>DASSAULT FALCON JET</td>
<td>FAA SAB536SW-D</td>
<td>FALCON 2000</td>
<td>Installation of the Battery Temperature System</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement, Report No. 36489-1, dated 8 August 1996: later revisions of the above listed documents approved by EASA in accordance with EASA in accordance with EASA 60/101-1 (or subsequent revisions of this document) and/ or the Technical Implementation Procedures of EU/USA Bilateral Agreement.</td>
<td>Applicability is limited to F2000 S/N 025. Prior to installation of this design change, it must be determined that the installation of this design change will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/11/2015</td>
<td>Active</td>
<td>1005604</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10055608</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 1</td>
<td>-</td>
<td>-</td>
<td>Installation of Night Vision Imaging System (NVIS) on E135/130/146- Side Cockpit with conventional alternative environment requirements. The following paragraph(s) at a later amendment: CS:27 Amdt. 3</td>
<td>Rotorcraft Flight Manual Supplement AE14131/RFMS Revision 1</td>
<td>Master Data List AE14131MDL Revision 1 for Night Vision Goggles (NVG) model in accordance with Master Data List AE1413/MDL.</td>
<td>Limited to helicopters with COPAS cockpit variant. The final installation of this kit must be supervised by the STC holder.</td>
<td>09/05/2020</td>
<td>Active</td>
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<tr>
<td>10055615</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>-</td>
<td>-</td>
<td>-</td>
<td>ACAS II Version 7.1 Upgrade</td>
<td>CS:25 Amdt. 17, IAW Certification Plan S3401-CLP01 Ed1 for additional environmental protection and emissions levels of the original product unchanged and remain applicable to this certificate approval.</td>
<td>S3401-HDK01 Flight Manual Supplement Ed3 or later revision of the above listed documents approved by EASA.</td>
<td>None</td>
<td>23/11/2015</td>
<td>Active</td>
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<tr>
<td>10055631</td>
<td>CRANFIELD AEROSPACE LTD</td>
<td>-</td>
<td>-</td>
<td>-</td>
<td>TCAS II Upgrade to Version 7.1</td>
<td>Cranfield Aerospace AFM Supplement CA/DD/M043</td>
<td>Applicable to aircraft serial number 614 only. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce noise effect upon the airworthiness of the product.</td>
<td>24/11/2015</td>
<td>Active</td>
<td>10056631</td>
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<tr>
<td>10055632</td>
<td>WELENS ENGINEERING COMPANY Inc.</td>
<td>FAA</td>
<td>SAU03480 REV. 4</td>
<td>-</td>
<td>Installation of Parmetheus Plus LED PAR36 and PAR46 taxi and landings lights.</td>
<td>Whelen STC Manual, Form 14973, dated July 7, 2014.</td>
<td>Whelen STC Manual, Form 14973, dated July 7, 2014. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce noise effect upon the airworthiness of the product.</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10056632</td>
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<tr>
<td>10056633</td>
<td>AIRCRAFT TECHNICAL DEVELOPMENT Inc. do/b AEROCON ENGINEERING COMPANY</td>
<td>FAA</td>
<td>STO2545LA</td>
<td>-</td>
<td>Installation of Cabin Interior Noise Reduction Kit.</td>
<td>Aeroson Engineering Company Master Data List AC25-NRB741-10000 Revision &quot;0&quot; dated 03/08/2015, Aeroson Engineering Company Instructions for Continued Airworthiness Document Maintenance Manual Supplement MWS-B747-1818 Revision &quot;NC&quot; dated 14/01/2014.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce noise effect upon the airworthiness of the product.</td>
<td>24/11/2015</td>
<td>Active</td>
<td>10056633</td>
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<td>10056637</td>
<td>SABENA TECHNICS BOD</td>
<td>-</td>
<td>-</td>
<td>-</td>
<td>BBJ Turnkey Taiwan Eagle conversion VIP</td>
<td>BBJ Turnkey Taiwan Eagle conversion VIP</td>
<td>- Only applicable for MSN 61040- As per Aircraft Flight Manual Supplement, Ref. AFMS-B3031-00-031 Issue B, Section 6 - Limitations. As per Instruction for Continued Airworthiness Ref. ICA-B3031-00-032 Issue B, Section 4 - Airworthiness Limitations. Prior to installation of this design change it must...</td>
<td>25/11/2015</td>
<td>Active</td>
<td>10056637</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>I.S.E.I INTERNATIONAL SERVICO ELECTRONIQUE INFORMATIQUE</td>
<td></td>
<td></td>
<td>SR22, SR22T</td>
<td>Installation of SAFETYPLANE V5 System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document Ref. SSTD - 44-15/02M Issue 03 dated 24/09/2015 Instruction for Continued Airworthiness Ref. A-00-15-63-AFM Rev. 1.0 - Supplemental Type Definition Document Ref. SSTD - 44-15/02M Issue 03 dated 24/09/2015 Instruction for Continued Airworthiness Ref. A-00-15-63-AFM Rev. 1.0</td>
<td>None</td>
<td>26/11/2015</td>
<td>Active 10205022</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td></td>
<td>MOD 15181 - TCAS Change 7.1</td>
<td>1900C, 1900D</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document Ref. SSTD - 44-15/02M Issue 03 dated 24/09/2015 Instruction for Continued Airworthiness Ref. A-00-15-63-AFM Rev. 1.0 - Supplemental Type Definition Document Ref. SSTD - 44-15/02M Issue 03 dated 24/09/2015 Instruction for Continued Airworthiness Ref. A-00-15-63-AFM Rev. 1.0</td>
<td>None</td>
<td>26/11/2015</td>
<td>Active 10205064</td>
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<td>STC TWENTY ONE LIMITED</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>767-200</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document Ref. SSTD - 44-15/02M Issue 03 dated 24/09/2015 Instruction for Continued Airworthiness Ref. A-00-15-63-AFM Rev. 1.0 - Supplemental Type Definition Document Ref. SSTD - 44-15/02M Issue 03 dated 24/09/2015 Instruction for Continued Airworthiness Ref. A-00-15-63-AFM Rev. 1.0</td>
<td>None</td>
<td>27/11/2015</td>
<td>Active 10205063</td>
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<tr>
<td>MECAR AVIATION GROUP S.p.A.</td>
<td>REV. 4</td>
<td></td>
<td>A1W189</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document Ref. SSTD - 44-15/02M Issue 03 dated 24/09/2015 Instruction for Continued Airworthiness Ref. A-00-15-63-AFM Rev. 1.0 - Supplemental Type Definition Document Ref. SSTD - 44-15/02M Issue 03 dated 24/09/2015 Instruction for Continued Airworthiness Ref. A-00-15-63-AFM Rev. 1.0</td>
<td>None</td>
<td>07/06/2023</td>
<td>Active 10205036</td>
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**Notes:**
- Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.
- List of EASA Supplemental Type Certificates.
- Further revisions of the above listed documents approved by EASA.
<table>
<thead>
<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>1005569</td>
<td>LEAT S.p.A.</td>
<td></td>
<td></td>
<td>CITATION II</td>
<td>TCAS II VERSION 7.0 MODIFICATION/Twa Pre mod configuration consists in TCAS II version 7.0 installation processor TPU 57APN 086-01146-1211. Post mod configurations consist in TCAS II version 7.1 installation processor TPU 67B PN 086-01146-2221.</td>
<td>Aircraft Flight Manual Supplement n. 34-40-015-0600 Rev. 0 dated 05.11.2015 or later revisions of the above listed documents approved by EASA Master Data List DRW N. 600-0904-600-000 Rev. 07 dated 06.11.2015CA TAS II Version 7.1 Modification 34-40-015-0600 rev. 00 dated 05.11.2015</td>
<td>Limited to Aeroplanes serial No. 550-0393 and 550-0081. This STC can be installed provided STC EASA A.5.0043 has been installed.</td>
<td>27/11/2015</td>
<td>Active</td>
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<td>10055724</td>
<td>WHELEN ENGINEERING COMPANY Inc.</td>
<td></td>
<td>FAA STC SRO006BO</td>
<td>SEE FAA AMM-150212 REV --</td>
<td>Installation of Parafmepus Plus PAR36 and PAR46 taxi and landing lights.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>See FAA AMM - 150212, Revision -- dated 10.09.2015</td>
<td>Prior to installation of this design change it must be determined that no adverse effect upon the airworthiness of the product.</td>
<td>09/12/2015</td>
<td>Active</td>
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<td>10055726</td>
<td>ROSEMOVNT AEROSPACE, Inc.</td>
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<td>FAA STC ST0171CH</td>
<td>777-300 ER 777-300 777-200</td>
<td>Installation of a Cockpit Door Surveillance System (CDSS).</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, Rosemount Aerospace document No. D2030682, Revision 19 dated 14 November 2011.; - Inverter/Pump Installation dated November 28, 1988.; - Installation and Operating Instructions, Form EA 109K dated November 30, 2015.</td>
<td>None</td>
<td>30/11/2015</td>
<td>Active</td>
<td>10055726</td>
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<tr>
<td>10055735</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>FAA STC SA1123QL</td>
<td>MYSTERE-FALCON 900</td>
<td>Upgrade TCAS 2000 to Change 7.1 on M9000</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Flight Manual Supplement, DFS Supplement N°36, Original Issued 18 November 2015 or later revisions of the above listed documents approved by EASA Statement of Compliance DFS-DE-24- F9-4774 Issue A dated November 10, 2015</td>
<td>Applicability is limited to M9000 model. Prior to installation of this design change it must be determined that no adverse effect upon the airworthiness of the product.</td>
<td>30/11/2015</td>
<td>Active</td>
<td>10055735</td>
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<tr>
<td>10055739</td>
<td>AERO MEDICAL PRODUCTS MFG. Inc.</td>
<td></td>
<td>FAA STC SA1123QL</td>
<td>B200GT, B200T, B300, B300C, B200C, B200COT, B200CT, A100-1, A200, A200C, A200CT 200CT, 200T, 300, 300LW 1900, 1900C, 1900D, 200, 200C</td>
<td>- Install Aero Medical Products Mfg., Inc Air Ambulance Stretchers PN7125 using Installation Kit PN 805. - The Medical Oxygen System Kit PN 100 is an FAA approved option. - The Inverter Pump Kit No. 812 may be installed.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Installation and Operating Instructions, Form EA 109K dated November 28, 1988.; - Installation and Operating Instructions, EA Form 170, dated December 2, 1991.</td>
<td>1. FAA Approved Aircraft Flight Manual Supplement dated July 2, 2001 (required). 2. Prior to the first flight of each modified aircraft, and after each change to the above list of documents approved by EASA Statement of Compliance DFS-DE-24- F9-4774 Issue A dated November 10, 2015.</td>
<td>09/12/2015</td>
<td>Active</td>
<td>10055739</td>
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<tr>
<td>10055745</td>
<td>32B SUPPORT SERVICES GmbH</td>
<td></td>
<td>REV. 1</td>
<td>DORNIER 228-212 DORNIER 228-302 DORNIER 228-201 DORNIER 228-200 DORNIER 228-101 DORNIER 228-100</td>
<td>Approval of TCAS Change 7.1 software upgrade.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with 328 Design modification DC-12403220001/A dated 25.11.2015 and MDL ref MDL0357201 Issue A, dated 25.11.2015. Operate in accordance with 328 Design Flight Manual Supplement n0AMRA551034/420001 Rev A, dated 21.11.2015 or later revisions of the above listed documents.</td>
<td>Prior to installation of this design change it must be determined that no adverse effect upon the airworthiness of the product.</td>
<td>27/06/2018</td>
<td>Active</td>
<td>10055745</td>
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<td>EASA STC No.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>STC Ref.</td>
<td>System Record</td>
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<td>1005574</td>
<td>LEARJET Inc.</td>
<td>REV. 2</td>
<td>Active</td>
<td>FAA STC ST/110329C</td>
<td>Installation of a wireless cabin telecommunications router</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments.</td>
<td>1. Future Jet Aerodesign master document No. 193-34700-MLD-Revision 8, dated 20 September 2010, or later approved FAA revision.</td>
<td>The installer must notify FAA if all documents contained in the STC and FAA STC document 193-34700-AMR are not in compliance, as of 20 September 2010, or later approved FAA approval.</td>
<td>15/11/2019</td>
<td>FAA-81102</td>
<td>1056574</td>
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<td>1005574</td>
<td>POLYTECH AERO SUPPORT INC.</td>
<td>REV. 1</td>
<td>Active</td>
<td>TC/OT SA15-88 ISSUE 1</td>
<td>Provisions for B737 EFB Mount and Power Connection Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product have been unchanged and remain applicable to this certificate/approval.</td>
<td>Modification Summary, Ref. MS-15-005, Revision 0, dated October 8, 2015 Instructions for Continued Airworthiness, Ref. ICA-15-005, Revision 0, dated October 8, 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise or adverse effect upon the airworthiness of the product.</td>
<td>11/12/2015</td>
<td>FAA-81102</td>
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<td>1005575</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 1</td>
<td>Active</td>
<td>FAA ST/20932CH-D</td>
<td>CABIN RECONFIGURATION WITH IFE eXlite System on A330</td>
<td>The Certification Basis of the original product remains applicable to this certificate/approval.</td>
<td>Master Data List, ref: 4802-02-A-MDL-800</td>
<td>According to the design related to this STC, this modification must be installed in conjunction with STC 10114507 rev.2 (or later revision). NonElectric Floor Path Marking System Prior to installation of this design change it must be determined that the interrelationship between this design change and...</td>
<td>01/12/2015</td>
<td>FAA-81102</td>
<td>1056575</td>
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<td>1005577</td>
<td>STANDARD AERO</td>
<td>REV. 1</td>
<td>Active</td>
<td>FAA ST/20932CH-D</td>
<td>Installation of an AirCell Access Iridium Phone System within True North/Express WiFi System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product have been unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List 1019567 Rev D dated Feb 03, 2011</td>
<td>As per AFMS A2, the wireless LAN are not to be utilized on the flight deck. This system is intended to provide internet access and...</td>
<td>04/03/2016</td>
<td>FAA-81102</td>
<td>1056577</td>
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<td>1005577</td>
<td>P3 ENGINEERING GMBH</td>
<td>REV. 1</td>
<td>Active</td>
<td>FAA ST/20932CH-D</td>
<td>AIRCELL SBB SATCOM AND WIFI CABIN NETWORK SYSTEM INSTALLATION</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product have been unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Program / Compliance Summary, ref: doc.PV-00034-CPR-0001-D, dated 11/0 November 2015 - Master Document, ref: doc. P5- AA01-MDL-0001, issued B, dated 26th September 2018 &amp; Date of Issue for C...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise or adverse effect upon the airworthiness of the product.</td>
<td>25/10/2018</td>
<td>FAA-81102</td>
<td>1056577</td>
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The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

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<th>Foreign/NA A STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10055886</td>
<td>AERO-DENIST GMBH &amp; CO. KG</td>
<td></td>
<td></td>
<td>HS 125 SERIES 700B</td>
<td>TCAS TTR-4000 Upgrade to Version 7.1</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements FAR 25 Amendment 20-G, FAR 25.29, FAR 25.1301, FAR 25.1309; FAR 25.1331; FAR 25.1353; FAR 25.1357; FAR 25.1413; FAR 25.1519; FAR 25.1519; FAR 25.1581; FAR 25.1583; FAR 25.1583; FAR 25.1583...</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/12/2015</td>
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<td>10052020</td>
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<td>10055876</td>
<td>JET AVIGATION ST. LOUIS, Inc.</td>
<td>FAA STC</td>
<td>ST04036CH</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Installation of a Window Shade System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Aero-Denist Avion Supplement AG1005-071 Rev.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/12/2015</td>
<td>Active</td>
<td>10052016</td>
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<td>10055877</td>
<td>SIRIUS AEROTECH S.L.</td>
<td>FAA STC</td>
<td></td>
<td>MYSTERE-FALCON 900</td>
<td>Change from TCAS 7.0 to TCAS 7.1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>JET Aviation St. Louis, Inc. Master Data List 34-157G001 Rev. A, dated October 23, 2015</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/12/2015</td>
<td>Active</td>
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<td>10055883</td>
<td>SABENA TECHNICS BOD</td>
<td>FAA STC</td>
<td>ST04036CH</td>
<td>A340-300 (A340-313)</td>
<td>AIR TAHITI NUI - CABIN AND IFE MODIFICATION/Cabin configuration change from 6 F/C-24 B/C / 264Y/C to 32 B/C / 264Y/C with new B/C STELIA Erosito II seats and Y/C CEVEN Swayy seats.Installation of new ZODIAC Innovations RAVE IFE system with AVOD seats,Evolution II seats, and Y/C GEVEN Steezy B/C / 264Y/C with new B/C STELIA A340-300 (A340-313)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List. TCAS 7.1 Upgrade F500, SA15-375-MDL, rev. 0;FLIGHT MANUAL SUPPLEMENT, TCAS 7.1 Upgrade F500, SA15-375-AFM, rev. for later revisions of the above listed documents approved by EASA.</td>
<td>As per the LIMITATIONS section in the FLIGHT MANUAL SUPPLEMENT, SA15-375-AFM, rev. 0, Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/12/2015</td>
<td>Active</td>
<td>10052083</td>
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<td>10055885</td>
<td>ASSOCIATED AIR CENTER, L.P.</td>
<td>FAA STC</td>
<td>ST111115C</td>
<td>747-8</td>
<td>Provisions for the Installation of the Civil Aircraft Missile Protection System (CAMPS)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Document List, Ref. 3129000000. Rev IR.Instructions for Continuous Airworthiness, Ref. 3129000000. Rev IR. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/12/2015</td>
<td>Active</td>
<td>10052080</td>
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<td>10055904</td>
<td>JET AVIGATION ST. LOUIS, Inc.</td>
<td>FAA STC</td>
<td>ST03444CH</td>
<td>BD-700-1A10</td>
<td>Modification of an Executive Interior Arrangement Including Cabin Seats, Divans, Mid Cabin Pocket Door, Removable Mid Cabin Bulkhead, Passenger/Oxygen System Mods, Emergency Light System Mods and Emergency Equipment.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>JET Aviation St. Louis, Inc. Master Data List 35-299G041 Rev. A, dated April 22, 2015. The FAA approved AFMS document defined by this Master Data List is required.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/12/2015</td>
<td>Active</td>
<td>10052004</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>One-time Modification of an Executive Interior Arrangement including Cabin Seats, Divans, Mid-Cabin Pocket Door, Removable Mid-Cabin bulkhead, Passenger Oxygen System Mods, Emergency Lighting System Mods and Emergency Equipment.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Jet Aviation ST. Louis Inc. Master Data List 35-2000/040 Rev. A; dated February 7, 2015. The FAA approved AFMS document defined by the Master Data List is required.</td>
<td>1) This approval is limited to only the installation made in Bombardier’s, model BD-700-1A10 serial number 9133. This STC does not permit manufacturing of parts for multiple installations. 2) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/12/2015</td>
<td>Active</td>
<td>ST005911</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>Installation of Satcom Direct Router (SDR) on Gulfstream GVI airplanes.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- FAA STC ST04027AT-D Certificate-GDL10000019 Index List Rev H; GCD000020829, Rev.-, EASA Validated Configurations List ST04027AT-D-CE61429M001 AFMS Rev B- CE61429M002 ICA Rev Dor later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical I...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/10/2021</td>
<td>Active</td>
<td>ST005913</td>
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<tr>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>Installation of Garmin GT5 855 TCAS I with dual antennas Garmin GAA58</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 423.301, 303, 305, 307, 365, 561,571, 576, 603, 607, 611, 867, 1301,1306, 1308, 1309, 1327, 1351, 1357, 1366, 1431, 1529, 1581, 1583, 1585, The...</td>
<td>Flight Manual Supplement AGR-2015-111-PMB-05-00, Amdt. 00; later revisions of the above listed documents approved by EASA - Instructions for Installation AGR-2015-111-97-10-00, Amdt. 00-Instructions for Continued Airworthiness AGR-2015-111-ICA-11-01 (Rev-01) dated 1st June 2017</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/06/2017</td>
<td>Active</td>
<td>ST005917</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>Installation of the Cockpit Voice Recorder System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Type- 4100S: Airplane Flight Manual Supplement Report No. 36391-1 Rev-A, dated 15 December 1995. Type – Alrred Signal: Airplane Flight Manual Supplement Report No. 36391-2, dated 10 October 1995 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004...</td>
<td>Applicability is limited to F2000 S/N 025.Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/12/2015</td>
<td>Active</td>
<td>ST005918</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>INTELSAT INFIGHT LLC</td>
<td>Installation of Gogo 2KU – Structural and Electrical Provisions/Ref 1: transfer of certificate to new holder Gogo LLC</td>
<td>STC Twenty One Ltd Instructions for Continued Amended/527, TEC-048 2Issue 1, Appendix A, Section 3 – Airworthiness Limitations/STC Twenty One Ltd Master Document List 521-MIL-1245</td>
<td>Only applicable to the following aircraft serial numbers: 26255 (&gt;)(&lt;) 29406 (&gt;)(&lt;) 30885 (&gt;)(&lt;) 32337 (&gt;)(&lt;) 32339 (&gt;)(&lt;) Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed equipment series P/ or repair will introduce noise effect upon the aircraft other product.</td>
<td>23/07/2020</td>
<td>Active</td>
<td>10100502</td>
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<tr>
<td>HONEYWELL INTERNATIONAL INC</td>
<td>Installation of the Honeywell TPA-100B TCAS II Processor with Change 7.1</td>
<td></td>
<td>None</td>
<td>10/12/2015</td>
<td>Active</td>
<td>10100044</td>
<td></td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>Installation of the Centralized Water System</td>
<td>Drawing List Report F24A-0717, Rev. 0 dated 18 November 1997</td>
<td>Applicability is limited to F2000 S/N 025. Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will introduce noise effect upon the aircraft other product.</td>
<td>10/12/2015</td>
<td>Active</td>
<td>10100047</td>
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<tr>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>Installation of a modular EMS-SARDIN™ equipment series P/ MBB-BK117 D-3 MBV-BK117 D-2</td>
<td>Initial Issue Master Data List, MOL-BK117 D-2-25-20-100,000-500, Rev. 0, dated 05 Oct2015 Certification Program, CR-BK117 D-2 (BK117 D-2-25-20-100,000-500), Rev. 1, Compliance Record, CR-BK117 D-2-25-20-100,000-500, Rev. 1, Flight Manual Supplement, FMS AKT78, Part I Rev. 0, Part II Rev. 0, Annex 1 Rev.</td>
<td>This STC is also applicable to the MBB-BK117 D-2 variant Minor Change 10070398 and 10070395 are prerequisites for the installation of Integration of AAT electrical system P/N 9030-525 cD-2 Minor Change 10070398 and 10070395 are prerequisites for the installation of Integration of AAT electrical system</td>
<td>30/11/2021</td>
<td>Active</td>
<td>10100048</td>
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<tr>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>Upgrade of Honeywell CAS-87A System to S/W-Change 7.1</td>
<td>Aircraft Flight Manual Supplement 12BE-APMS, Rev. N/Cor later revisions of the above listed documents approved by EASA, 12BE-CN, Rev. N, Change Note (Approval for Aircraft Modification) at aircraft approved revisions 12BE-ED, Rev. A, Engineering Order or later approved revisions</td>
<td>This STC does not cover aircraft equipped with Garmin G3000/AerionisSuite. Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will introduce noise effect upon the air...</td>
<td>16/12/2015</td>
<td>Active</td>
<td>10100054</td>
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<tr>
<td>ADAC LUFTHAFRT TECHNIK GmbH</td>
<td>Night Vision Imaging System (NVIS) follow-on installation. Certification of additional NVIS configuration as defined in Master Document ListEC145-04106-001-LMD-00.</td>
<td>RFSMS EC145-04106-025- MFS-001 later revisions of the above listed documents approved by EASA/EAC145-04106-027- RCS-00EC145-04106-001-LMD-00</td>
<td>Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will introduce noise effect upon the aircraft other product.</td>
<td>11/12/2015</td>
<td>Active</td>
<td>10100087</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>TELAIR INTERNATIONAL Inc.</td>
<td></td>
<td>FAA STC ST02540LA</td>
<td></td>
<td>Installation of Telair International LLC light weight PINs PMC 100 and PA9100 series cargo pallets.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. New-PAGE. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td></td>
<td>1. Only eligible for installation on transport category airplane as mentioned in the Approved Model List (AML) SA20540LA and that have been certified with a cargo restraint system that is compatible with the certification data sheet presented.</td>
<td>17/12/2015</td>
<td>Active</td>
<td>10055988</td>
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<tr>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td></td>
<td></td>
<td>737-800</td>
<td>Design Change ERT/CNG/742 - B737-800 Cabin reconfiguration: Passenger seat configuration change from 8C 154Y to 18BY.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Eirtech Aviation Services Master Document List MDL/742-009 Issue 6, Certification Programme CP742-003 Issue 7, Design Report DRV 742-020 Issue 6, Declaration of Compliance Issue 1, Change &amp; Repair Classification CRC/742-002, Equipment Assessment EA/742-005 Issue 3, Service Bulletin ERT-SB-803 Issue 4.</td>
<td>None</td>
<td>14/12/2015</td>
<td>Active</td>
<td>10055994</td>
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<tr>
<td>ABLE AEROSPACE, Inc.</td>
<td>REV.</td>
<td>FAA PMA PO24687NM</td>
<td>209L-3, 209L-4, 505 209L-3, 209L-4, 209A, 209B, 209L-1</td>
<td>EASA Approval of Able Aerospace PMA Strap Assembly P/N 302A-21-001/101/103 eligible for installation on Bell 206C-1 and 206L-4 aircraft as per FAA PMA No. PO24687NM Supplement No. 34, dated 13 December 2021. Rev. 1: Addition of Bell 505 eligibility to the EASA STC.</td>
<td>The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unachieved and remain applicable to this certificate/approval. New-PAGE. The requirements for environmental protection and the associated certified noise and/ or emissions remaining applicable to this certificate/approval.</td>
<td>Supplemental Instruction for Continued Amement Report4241-AL, Rev. NC, dated 27 January 2014 or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement Certification Compliance Re...</td>
<td>Prior to installation of this design change it must be determined that the relative change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/06/2023</td>
<td>Active</td>
<td>10056001</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA</td>
<td>SAB2415SW-D</td>
<td>F2000</td>
<td>Installation of the Collins Traffic Alert and Collision Avoidance System (TCAS)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Drawing List Report F2M-01013, Rev.A dated 13 April 1995</td>
<td>Applicability is limited to F2000 S/N 025. Prior to installation of this design change it must be considered that the relative change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/12/2015</td>
<td>Active</td>
<td>10056048</td>
<td></td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA</td>
<td>SAB235SW-D</td>
<td>F2000</td>
<td>Installation of the Antenna Installations</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Drawing List Report F2M-01002, Rev.C dated 01 May 1996</td>
<td>Applicability is limited to F2000 S/N 025. Prior to installation of this design change it must be considered that the relative change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/12/2015</td>
<td>Active</td>
<td>10056050</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA</td>
<td>SAB244SW-D</td>
<td>F2000</td>
<td>Installation of the Basic Interior Furnishings</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Drawing List Report F2M-01014, Rev.M dated 19 June 1996</td>
<td>Applicability is limited to F2000 S/N 025. Prior to installation of this design change it must be considered that the relative change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/12/2015</td>
<td>Active</td>
<td>10056051</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10056055</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
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<td>A300 CA-650 R VARIANT F</td>
<td>TCAS V7.1 Upgrade</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment CS25.1CS 25.1301CS 25.1302, CS 25.1309, CS 25.1316, CS 25.1317, CS 25.1322, CS 25.1351, CS 25.1353, CS 25.1357, CS 25.1431.</td>
<td>S4A Flight Manual Supplement A3005-HDK01 Issue 3 after later revisions of the above listed documents approved by EASA. The compliance documents detailed in S4A STC Major Change Declaration Of Compliance Form A3005-STCA01 Issue 1</td>
<td>Applicable only to A300 CA-650R variant F aircraft with ACSS T2CAS Computer Unit part number 9000000-11212 previously installed Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/12/2015</td>
<td>Active</td>
<td>10056055</td>
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<tr>
<td>10056059</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA</td>
<td>SA2405SW-D</td>
<td>F2000</td>
<td>Installation of the Miscellaneous Structural Modifications (Avionics/Electrical)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product airworthiness and remain applicable to this certificate/approval.</td>
<td>Drawing List Report F2M01096 Rev.H dated 28 April 1996</td>
<td>Applicability is limited to F2000 S/N 025 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/12/2015</td>
<td>Active</td>
<td>10056059</td>
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<td>10056062</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC</td>
<td>SR09609R</td>
<td>407</td>
<td>Installation of MD302 Standby Attitude Module Kit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product airworthiness and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement, Report No. AA-14091, No Revision, dated April 27, 2015, Master Drawing List, Report No. AA-14096, Rev. A, dated March 13, 2015 Instructions for Continued Airworthiness, Report No. AA-14088, Rev. A, dated April 27, 2015.</td>
<td>Eligible for installation on S/N 54300 and subsequent Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10056062</td>
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<tr>
<td>10056085</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>FAA STC</td>
<td>SR03173CH</td>
<td>501</td>
<td>AVIONIC MODIFICATION (Weather RADAR, TCAS, TAWS, FMS, GPS)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product airworthiness and remain applicable to this certificate/approval.</td>
<td>50003-CH0150003-HDK01 AFM-S90003-HDK02 MI-90/95 later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2015</td>
<td>Active</td>
<td>10056085</td>
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<tr>
<td>10056089</td>
<td>ISOLAIR Inc., ISOLAIR HELICOPTER SY</td>
<td>FAA STC</td>
<td>SR03173CH</td>
<td>F-28C, F-28F 280C, 280F, 280FX</td>
<td>Title Isolair Model 3900-280 Agricultural Spray System Description Installation of an externally mounted agricultural spray system consisting of belly mounted pod, spray boom, spray pump, tank, liquid contents jettison system and pilot controls</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product airworthiness and remain applicable to this certificate/approval.</td>
<td>EASA Rotorcraft Flight Manual Supplement for Enstrom Model 280 Series with the Isolair Model 3900-280 Spray System DFMS No.3900-280FMS-EASA, Initial Issuance later revisions of the above listed documents approved by EASA Master Drawing List for Isolair Model 3900-280 Spray System, MSL No. 3900-280Rev1.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2015</td>
<td>Active</td>
<td>10056089</td>
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<tr>
<td>10056093</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Installation of an Aviator 300 Swift Broadband System</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements FAR 25 Am1h 25-1 thru 25-7825.29, 25-391, 25-103, 25-305, 25-307, 25-321, 25-369, 25-561, 25-561, 25-601, 25-601, 25-603, 25-605, 25-607, 25-609, 25-611, 25-613, 25-625, 25-</td>
<td>STC Twenty One Ltd Flight Manual Supplement S21-FMS-0126 Issue 1STC Twenty One Ltd Instructions for Continued Airworthiness S21-TEC-0487 Issue 1 (&gt;)(&gt;) Section 3 – Airworthiness Limits or later revisions of the above listed documents approved by EASA. STC Twenty One Ltd Master Document List S21-M01...</td>
<td>1. Applicable to aircraft MSN 5980 and 5983 only. 2. This design change is only certified for the aircraft configuration at the time of this document issue. Any changes to the aircraft for the following reasons will require re-evaluation: a) The aircraft equipment fitted is appreciably different an...</td>
<td>17/12/2015</td>
<td>Active</td>
<td>10056093</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10056102</td>
<td>LEARJET Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST0172WI</td>
<td>45</td>
<td>Installation of a seven passenger floor plan (SA-3) on Learjet Model 45 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. FAA Master Document List M1055681CT, Revision (G), dated 20 December 2014. FAA approved 28 December 2014, earlier FAA approved version. 2. Instructions for Continued Airworthiness, M1055681CT-ICA-001, dated 02 October 2014. FAA accepted December 28, 2014.</td>
<td>14/06/2019</td>
<td>Active</td>
<td>10056102</td>
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<tr>
<td>10056110</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>737-600/-700/-800/-900-9</td>
<td>00ER</td>
<td>EFB Tablet Display Mount Provision</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Flight Manual Supplement, Refs 2082919-AM B Issue 2, 2082919-AM C Issue 2. Continued Airworthiness, Ref. 2082919-CA, Issue 1 or later revisions of the above listed documents approved by EASA Engineering Order and Certification Program, Ref. 2082919-EO, Issue 6.</td>
<td>As per AFMS Aircraft Flight Manual Supplement, Refs 2082919-AM B Issue 2, 2082919-AM C Issue 2 Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce noadverse...</td>
<td>18/12/2015</td>
<td>Active</td>
<td>10056110</td>
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<tr>
<td>10056115</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>A321-111, A321-212</td>
<td>Installation of Swiss Max cabin reconfiguration/ - Approval Extension to include A321-111.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental Type Design Definition STDD25-1021M Iss.1 – Rev. 03, dated 11/06/2016; Instruction s for Continued Airworthiness A-06-15/38-ICA Iss.1 – Rev. 02, dated 26/08/2015 Statement of Compliance Applicant Declaration of Compliance (ADoC) – net 3959, dated 07 Oct 2016 or later revisions of the above Iss.</td>
<td>This design approval shall be installed in conjunction with STC10051686, Rev. 1 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse...</td>
<td>10/10/2016</td>
<td>Active</td>
<td>10056115</td>
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<tr>
<td>10056116</td>
<td>SABENA TECHNIKS BOD</td>
<td>REV. 1</td>
<td>F28 MARK 0100</td>
<td>Ventral and Wings Pods - F28 Mark 0100Rev 1: Removal of Limitation to MSN 11290</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment CS25.858 at Amendment 3, Cargo or Baggage compartment smoke or fire detection systems The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>AFMS-FK104-00-016 issue CICA-FK104-00-017 issue ADeclaration of Compliance: MAS S-00-2113 Issue 8 TRS- S-00-2113 Issue 9 or later revisions of the above listed documents approved by EASA</td>
<td>For all configurations - The aircraft may not be converted back to a pax configuration AFMS-FK104-00-016 section 3 Limitations - The aircraft is not to be used in civil/commercial operations (nouclic commercial transport)</td>
<td>29/04/2020</td>
<td>Active</td>
<td>10056116</td>
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<td>10056117</td>
<td>VENTURA AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST0949LA</td>
<td>MD-88 DC-9-41, DC-9-82, DC-9-83</td>
<td>9G Cargo Barrier - Installation of a cargo barrier on the main deck in front of the fuselage on Boeing DC-9 and MD-88 aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>Chapter 12 of Report No. VA-9B-1150 Rev. II &quot;Maintenance and test procedures Continued Airworthiness Manual MD-80 Series Right Cargo Barrin Installation&quot;, Airworthiness Limitations Section 15.52, 15.54, 15.55 Later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (...</td>
<td>18/12/2015</td>
<td>Active</td>
<td>10056117</td>
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<td>STC Holder</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC 10056127</td>
<td>GV</td>
<td>Installation of a Future Air Navigation System (FANS). 1-Alt/AFN(&gt;)=...</td>
<td></td>
<td></td>
<td>Prior to installation of this design change it must be determined that there be...</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10056127</td>
<td></td>
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<tr>
<td>MED-PAC INC.</td>
<td>FAA STC 10056129</td>
<td></td>
<td>Installation of an air ambulance conversion kit.</td>
<td></td>
<td></td>
<td>Prior to installation of this design change it must be determined that there be...</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10056129</td>
<td></td>
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<tr>
<td>AEROLITE AG</td>
<td>REV. 4</td>
<td>A381</td>
<td>EMS Equipment PN 168-0404200-501 Revision 1 modifies the EMS Equipment PN 168-0404200-501 to include...</td>
<td></td>
<td></td>
<td>Pre-requisites for installation of PNs 168-0404200-501 are given in the Operations Manual OPM-168-0404200-501, F, dated 2019-03-15. For any installation onto the Cargo Platform a separate design approval is required (not scope of this STC). Prior to installation of this change/repair it must be approved, and for the permanent records for the change/repair it must be determined that there be...</td>
<td>22/05/2019</td>
<td>Active</td>
<td>10056138</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>A380-881, A380-842</td>
<td></td>
<td>PAC GCS A380 Large Antenna Installation</td>
<td></td>
<td></td>
<td>None</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10056134</td>
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<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>C900T1</td>
<td>Installation of Rockwell Collins, Inc. Pro Line Fusion ® cockpit.</td>
<td></td>
<td></td>
<td>(1) Approval of this change in type design applies to the above model/aircraft only; (2) Compatibility of this design change with previously approved modifications must be determined by the installer; (3) A copy of this Certificate must be maintained as part of the permanent records for the modified a...</td>
<td>10/02/2017</td>
<td>Active</td>
<td>10056137</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
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<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10058140</td>
<td>AIRBUS S.A.S.</td>
<td>Active</td>
<td>A319-133</td>
<td>Airbus Corporate Jet Centre Modification MOD CJ1930 “Top ModificationApollo – Cabin and Systems Upgrade”</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (a) paragraph (a) at a later amendment CS25.1302 amendment 37e the requirements for environmental protection and the associated certified noise and/or emissions level.</td>
<td>Modification Approval Sheet – MAS ref CJ1930, dated 18th-December 2015</td>
<td>This STC is limited to MSN 2192</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10058140</td>
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<tr>
<td>10058144</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>F A A</td>
<td>ST04033AT-D</td>
<td>GV</td>
<td>Installation of Gogo UCS 5000 Universal Cabin System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Gulfstream Aerospace Corporation Index List CES14282019 Rev ***, dated October 15, 2015. The Airplane Flight Manual Supplement, CES14285013 Revisions ***, dated October 15, 2015, is required and must be carried in the aircraft on all flights. The Instructions for Continued Airworthiness CES1428A013, Revision A. Use of the UCS 5000 Wi-Fi function by the Flight Crew whilst in the cockpit during flight is not permitted. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness.</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10058144</td>
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<tr>
<td>10058146</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA</td>
<td>ST04031AT-D</td>
<td>FALCON 2000</td>
<td>Flight Data Recorder System Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Drawing List Report No. F3M/01/24, Revision C. Airplane Flight Manual Supplement. Ref. 36590-1. Revision A. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness.</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10058146</td>
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<tr>
<td>10058147</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA</td>
<td>ST04034AT-D</td>
<td>G-IV</td>
<td>Installation of a UCS 5000 Aircraft Universal Cabin System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Gulfstream Aerospace Corporation Index List CE41428080 (v-3) Revisions ***, dated October 26, 2015. The Airplane Flight Manual Supplement CE41428005, Revision (v-3) dated October 26, 2015, is required and must be carried in the aircraft on all flights. The Instructions for Continued Airworthiness, CE41428A013, Revision A. Use of the UCS 5000 Wi-Fi function by the Flight Crew whilst in the cockpit during flight is not permitted. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness.</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10058147</td>
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<tr>
<td>10058150</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>HAWKER 800XP</td>
<td>Upgrade of Dual Honeywell Control Display UnitsExisting two (2) Honeywell Flight Management System Control Display Units (CD-830) are replaced with updated touchscreen technology units (CD-830)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: (a) paragraph (a) at a later amendment CS25.1302 amendment 37e the requirements for environmental protection and the associated certified noise and/or emissions level.</td>
<td>Design Organization QC Declaration of Compliance RH46-DG501-001-M0- DOCRevision 1 from 21 December 2015 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/12/2015</td>
<td>Active</td>
<td>10058150</td>
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<td>EASA STC no.</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>GULFSTREAM AVIATION BUSINESSSERVICES AGS</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST04021AT-D</td>
<td>GULFSTREAM 200 / GALAXY</td>
<td>Installation of ADS-B Out System</td>
<td>requiring the following or Design alterations to the original product.</td>
<td>14/03/2016</td>
<td>Active</td>
<td>10056161</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>ROCKWELL COLLINS TCAS VERSION 7-1 (ACAS II) UPGRADE OF PRIOR INSTALLATION</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce a mismatch in the equipment.</td>
<td>22/12/2015</td>
<td>Active</td>
<td>10056168</td>
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<tr>
<td>TAMARACK AEROSPACE GROUP, INC.</td>
<td>TAMARACK AEROSPACE GROUP, INC.</td>
<td>REV. 7</td>
<td>525(M2)/525A, 525B</td>
<td>Installation of Tamarack ATLAS to Cessna models 525, 525A and 525B with ATLS enhancements. This change incorporates corrective actions to the uninstalled condition referenced in EASA AD 2019-0086 and improvements to the ATLS system.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce a mismatch in the equipment.</td>
<td>24/10/2019</td>
<td>Active</td>
<td>10056170</td>
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<td>EASA STC no.</td>
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<td>Limitations/Conditions</td>
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<td>10056177</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>-</td>
<td>-</td>
<td>Provision For Dual Portable Electronic Flight Bag (EFB) in a Boeing 757 Series Aircraft</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: 25.21(15)</td>
<td>06/2010/STC O REP STATIC STRENGTH TEST REPORT issue 1 13.01.2015/STC O RGMT 100501 TABLET HOLDER ASSY issue D 13.01.2015/STC O RGMT 100501 TABLET HOLDER ASSY issue B 13.01.2015/STC O RGMT 100501 TABLET HOLDER ASSY issue A tested (generic) 25.321(15)</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/12/2015</td>
<td>Active</td>
<td>10056177</td>
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<td>10056180</td>
<td>SIKORSKY AIRCRAFT CORPORATION</td>
<td>FAA STC SR0263NY-1</td>
<td>D</td>
<td>Installation of a Honeywell 1239 ZINGHUM</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CFR 14 FAR Part 25.321(15)</td>
<td>Rotorcraft Flight Manual Supplement, Honeywell 1239 ZINGHUM For Sikorsky S-76 C Aircraft (s/n 760511 and subsequent)</td>
<td>Eligible s/n 760511 and up</td>
<td>23/12/2015</td>
<td>Active</td>
<td>10056180</td>
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<td>10056185</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>FAA STC ST04211AT</td>
<td></td>
<td>Installation of antennas for ELT (Emergency Locator Transmitter) system</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CFR14 FAR Part 25 Amendment 17.25(15) &amp; 25.321(15)</td>
<td>S4A Flight manual Supplement no. 60422-HKD01 Edition 15A Maintenance manual supplement no. 60422-HKD02 Edition 1</td>
<td>None</td>
<td>04/01/2016</td>
<td>Active</td>
<td>10056185</td>
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<td>10056190</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC ST04211AT</td>
<td></td>
<td>Replacement of the original equipment Nickel-Cadmium (NiCd) main batteries with Concorde Lead-Acid, main batteries along with associated hardware on EADS CASA CN-235, CN-235-100, CN-235-200, CN-235-300 and C-295 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirement for environmental protection and the associated certified noise and/or emissions leveled the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Drawing List 5-0502 Revision B, dated March 18, 2015.</td>
<td>FAA Approved Airplane Flight Manual Supplement for the replacement with Concorde Lead-Acid, main batteries, 05-0504, Revision A, dated June 12, 2015, is a required part of the STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/01/2016</td>
<td>Active</td>
<td>10056190</td>
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<td>10056191</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC ST04185AT</td>
<td></td>
<td>Replacement of the original equipment ZingHUM valve regulated sealed Lead-acid batteries with associated hardware on the Dassault Falcon 7X series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirement for environmental protection and the associated certified noise and/or emissions leveled the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Drawing List 05-0336, Revision A, dated August 12, 2014.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/01/2016</td>
<td>Active</td>
<td>10056191</td>
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<td>STC Holder</td>
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<td>Model</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>TECH-TOOL PLASTICS, Inc.</td>
<td>FAA STC SR06947RC</td>
<td>AS 350 A, AS 350 D1, AS 350 B1, AS 350 B2, AS 350 B3, AS 350 BA</td>
<td>Installation of Bubble Windows in left hand or right hand side crewdoors.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval NEW-PAGE The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Rotorcraft Flight Manual Supplement (RFMS), Tech Tool Plastics, Inc. Document No. TTP-350-FMS-001. Revision IR, dated September 8, 2014. Issued Tech Tool Plastics, Inc. Master Drawing List. Document No. TTP-350-MLD-001, Rev A, dated July 10, 2014. Instructions for Continued Airworthiness (IAC).</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/echo effect upon the airworthiness of the product.</td>
<td>09/12/2016</td>
<td>Active</td>
<td>1056192</td>
<td></td>
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<tr>
<td>TECH-TOOL PLASTICS, Inc.</td>
<td>FAA STC SR95586HC</td>
<td>R44, R44 C1</td>
<td>Installation of Tech Tool Plastics replacement doors with full view windows.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Rotorcraft Flight Manual Supplement, Document No. TTP-RMS-002, Rev. IR, dated March 26&lt;sup&gt;th&lt;/sup&gt; 2016 is required. Rotorcraft Flight Manual Supplement, Document No. TTP-RMS-001, Rev. 0, dated March 2, 2010. Instructions for Continued Airworthiness, Document No. TTP-ICA-0035, Revision New, 01/04/2016.</td>
<td>The following R44-101 Forward Door Configurations are approved: R44-1-1, -2, -3, -4, -5, -6, -7, -8, -9, and -10.</td>
<td>05/01/2016</td>
<td>Active</td>
<td>1056193</td>
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<tr>
<td>AVONIR STRAUBING ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td>TBM700 N</td>
<td>Installation of multiple Antennas</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 4-301, 303, 305, 307, 369, 561-371, 575, 603, 607, 611, 625, 843, 867, 1301, 1306, 1308, 1309, 1327, 1351, 1341, 1529. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>- Project Description ASR-2015-077-PRD-04-00, Amendment 00. Instructions for Installation ASR-2015-077-IF-10-00, Amendment 00. Instructions for Continued Airworthiness ASR-2016-077-ICA-11-02 (Rev.02) dated 31st July 2017.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/echo effect upon the airworthiness of the product.</td>
<td>30/08/2017</td>
<td>Active</td>
<td>1056194</td>
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<tr>
<td>RHEINLAND AIR SERVICE GmbH</td>
<td>REV. 1</td>
<td>TBM700 N</td>
<td>Installation of Skywatch Sky497</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23, Amendment 4-301, 303, 305, 307, 369, 561-371, 575, 603, 607, 611, 625, 843, 867, 1301, 1306, 1308, 1309, 1327, 1351, 1341, 1385, 1381, 1431, 1531, 1...</td>
<td>- Flight Manual Supplement G800 RAS-15-01-02-FMS-00, Rev. 0 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/echo effect upon the airworthiness of the product.</td>
<td>05/01/2016</td>
<td>Active</td>
<td>1056195</td>
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<td>FOKKER SERVICES B.V.</td>
<td>REV. 1</td>
<td>CL-600-2E5 (RU SERIES 1500)</td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification Plan CP-0285 issue 3 Certification Plan CP-0286 issue 1 Compliance Record Report CRR-0285 issue 1</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/echo effect upon the airworthiness of the product.</td>
<td>06/01/2016</td>
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<td>EASA STC no.</td>
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<td>Foreign/NAA STC ref.</td>
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<td>10056206</td>
<td>WELENS ENGINEERING COMPANY Inc.</td>
<td>FAA SAMX00A</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Whelen Anti-Collision Position, and Anti-Collision Position Light System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Whelen Anti-Collision Light Systems Installation and Service Manual Document No. 05131, Rev. D dated May 2015 or later approved revision.</td>
<td>The installation must be checked in accordance with Whelen Anti-Collision Light Systems Installation and Service Manual Document No. 05131, Rev. D dated May 2015 or later approved revision install the following placards: Whelen Part Nos. 4421-1, or other approved equivalent, on view of the pilothigher OFF...</td>
<td>08/12/2016</td>
<td>Active</td>
<td>10056206</td>
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<tr>
<td>10056217</td>
<td>MARSHALL AVIATION SERVICES LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BD-700-1A10</td>
<td>To introduce role change operation for a Model Cabin Sliding Door currently installed on Bombardier Global Express (BD-700-1A10) aircraft MMSN 9506&lt;().&gt; registration OE-IVG</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Marshall Aviation Services Ltd Aircraft Flight Manual Supplement No. CS700-1 Supplement 1 Issue 1or later revisions of the above listed documents approved by EASA/Marshall Aviation Services Ltd Change Sheet No. SC0547A Issue 2.</td>
<td>Restricted to ASN 9811 A prerequisite is that EASA minor modification EASA.C.08522 installed on the aircraft.</td>
<td>07/01/2016</td>
<td>Active</td>
<td>10056217</td>
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<td>10056236</td>
<td>E.A.D AEROSPACE AIRWORTHINESS</td>
<td>777F 777-300ER 777-200LR 777-200</td>
<td>TCAS II Change 7.1 installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, EAD Aerospace document No. 01468-EAD-A-MLD, Revision C, dated 4 November 2015. - Airplane Flight Manual Supplement, EAD Aerospace document No. 01468-EAD-A-APM-A, Revision A, dated 14 October 2015or for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/01/2016</td>
<td>Active</td>
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<td>10056241</td>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>FAA STC ST04217AT</td>
<td>767-300</td>
<td>Interior Reconfiguration</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25 Amdt. 11: 25.1.25.2(a)(3), 25.301, 25.303, 25.601, 25.571(a)(b), 25.809, 25.813(a)(b), 25.815, 25.819, 25.791, 25.793, 25.815, 25.815(b)(a) (5)(V)(7)(8), 25.853(a)(k)(i)O(i), 25.869, ...</td>
<td>HAECO Master Data List 14T2056L01 rev BHAECO Instructions for continued airworthiness. 14T205R009 rev BHAECO Weight And Balance report 14T205R017 rev Aor later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/04/CF (or subsequent revisions of this document).</td>
<td>STC Aerospace Photoluminescent floor proximity emergency escape pathmarking system (EASA STC 10015434/FAA STC 702051A) must be installed prior to or concurrently with this STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and...</td>
<td>08/01/2016</td>
<td>Active</td>
<td>10056241</td>
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<tr>
<td>10056242</td>
<td>BOND HELICOPTERS EUROPE Ltd</td>
<td>REV. 1</td>
<td>FAA STC ST04217AT</td>
<td>EC135 T2+</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at or later amendment: CS 27 Amdt 37 CS 27 Amdt 3 Appendix ACS 27 Amdt 3 Appendix ACS 27 Amdt 3 Appendix ACS ACSN Initial Issue ACNS.B.5.VOS.001,A,...</td>
<td>Flight Manual Supplement for NPAS Md Life Upgrade, EC135 (includingGarmin GTN 650). FMS/104-002 Issue 4, dated 7 January 2016 or later revisions of the above listed documents approved by EASA. - Design Report DR-104-001, Issue 5, dated 7 January 2016 - Compliance Check List CCL/104-003, Issue 5, dia...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/01/2016</td>
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<tr>
<td>10056256</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SR06539RC</td>
<td>407</td>
<td>Installation of a Polycarbonate Windshield Kit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List, Report No. AA-10052, Revision H, dated December 11.2013.Instructions for Continued Airworthiness, Report No. AA-10057, Revision A, dated January 20, 2014.</td>
<td>Polycarbonate Windshield kit is a replacement for the acrylic windshields for Bell Helicopter Textron Canada Limited Model 407 Helicopters. The Polycarbonate Windshield kit has been evaluated for compatibility with Night Vision Imaging Systems (NVIS). Prior to installation of this design change it...</td>
<td>13/01/2016</td>
<td>Active</td>
<td>10056256</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
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<tr>
<td>10056257</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SR10600AT</td>
<td>407</td>
<td></td>
<td>Manufacture of Hat Rack Safety Net.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental and protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Annotated Aircraft, Inc. Drawing List Report No. AA-98021, revision, dated February 5, 1998 and installation per Report No. AA-98020, no revision, dated February 5, 1998 or later approved revisions.</td>
<td>13/02/2016</td>
<td>Active</td>
<td>10056257</td>
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<tr>
<td>10056273</td>
<td>AVIATION SPECIALTIES UNLIMITED, INC</td>
<td>REV. 1</td>
<td>FAA STC SR017175E</td>
<td>SA 365 N, SA 365 N1 EC 155 B1 AS 365 N2, AS 365 N3</td>
<td>Installation of a supplemental lighting system - NVIS Lighting System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental and protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>ASU Flight Manual Supplement FMS-365004-8 Revision H, dated 01/05/2015; approved by FAA on 14/12/2015 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Proceeding...</td>
<td>01/03/2018</td>
<td>Active</td>
<td>10056273</td>
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<tr>
<td>10056275</td>
<td>HELICOPTER AIR TRANSPORT GESELLSCHAFT B. H.</td>
<td>REV. 5</td>
<td>EC135 T2+ EC135 T3 EC135 T1, EC135 T2 EC135 P2+ EC135 P3 EC135 P1, EC135 P2</td>
<td>Installation of Northern Airborne Technology Digital Audio Control System (n-DACS) in accordance with Helicopter Air Transport GmbH/Engineering Order EO 23-012 Installation of Northern Airborne Technology Digital Audio Control System (n-DACS) in accordance with Helicopter Air Transport GmbH/Engineering...</td>
<td>The Certification Basis for the original product as amended by this additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS27 Add. 3:27.21, 27.25, 27.27, 27.29, 27.301, 27.303, 27.305, 27.561, 27.603, 27.609, 27.610, 27.771, 27.1301, 27.13092.</td>
<td>Aircraft Flight Manual Supplement AFM 23-012 Revision 2 Certification Plan 23-012 Revision 3 (Order EO 23-012 Revision 4 Certification Report OR 23-012 Revision 4- Installation Manual III 23-012 Revision 4 Instructions for Continued Airworthiness ICA 23-012 Revision 4 for later revisions of the above.</td>
<td>Applicable to CDS, CPDS and Helionix cockpit variants. Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/07/2020</td>
<td>Active</td>
<td>10056275</td>
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<td>10056335</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 3</td>
<td>FAA SAC1754W1 B300C B300</td>
<td>Installation of Rockwell Collins, Inc. Pro Line Fusion® cockpit.</td>
<td>The Certification Basis for the original product as amended by this additional or alternative airworthiness requirements Special Condition (s): See EASA approved Amendment Equivalent Safety Finding (s): See EASA approved Amendment. The requirements for environmental protection and the associated certified...</td>
<td>Installation of Rockwell Collins, Inc. Pro Line Fusion® in accordance with Master Drawing List RCD4430 Revision (-), dated May 4, 2015.</td>
<td>(1) Approval of this change in type design applies to the above listed models only (2) Compatibility of this design change with previously approved modifications must be determined by the installer (3) A copy of this Certificate must be maintained as part of the permanent records for the modified aircraft.</td>
<td>20/08/2018</td>
<td>Active</td>
<td>10056335</td>
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<td>10056336</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>REV. 1</td>
<td>EC 130 T2</td>
<td>Installation of a modular &quot;quick conversion&quot; Air Ambulance Equipment configuration EC130-25-20-41-000-601 and EC130-25-20-41-000-606, in accordance with Master Data List MDL-EC130T2-EM5 Configuration EC130-25-20-41-000-601 incorporates a single stretchers installed on a structure called basic unit att...</td>
<td>The Certification Basis for the original product as amended by this additional or alternative airworthiness requirements JAR/CS-27.863 first issue as documented in CRI F01 &quot;Oxygen fire hazard in rotorcraft with Emergency Medical Service cabin configuration&quot;. The requirements for environmental...</td>
<td>Flight Manual Supplement FMS-AT6ES Rev. 01 Airworthiness Limitations Section as per section 6 of CML-EC130-AAT6ES Rev 02 later revisions of the above listed documents approved by EASA - Instruction for Continued Airworthiness (c)(c) OMEL-EC130-AAT6ES Rev 02 EMGS Operation Manual, OM-EC130T2-AAT01, ...</td>
<td>&quot;Quick conversion&quot; Air Ambulance Equipment configuration EC130-25-20-41-000-601 incorporates provisions (storage space, attachments and electrical power supply) for medical devices installation and operation of those devices are not part of this STC. Prior to installation of this design change it must...</td>
<td>17/08/2016</td>
<td>Active</td>
<td>10056336</td>
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The Certification Basis for the original product as amended/modified additional or alternative airworthiness requirements - Special Condition (s): See EASA approved Annex - Equivalent Safety Findings(s): See EASA approved Annex. The requirements for environmental protection and the associated/required change/repair it must be determined by the installer (2) Compatibility of this design change with previously approved modifications must be determined by the installer (3) A copy of this Certificate must be maintained as part of the permanent records for the modified aircraft.


Installation of Rockwell Collins, Inc. Pro Line Fusion® in accordance with Master Drawing List RCA-0430, Revision (-), dated December 19, 2014.Operation in accordance with (1) Airplane Flight Manual Supplement B200GT, B200CGT, RCA-0428, Revision (-), dated December 15, 2014; for (1) Airplane Flight Manual Supplement B200GT, B200CGT, RCA-0428, Revision (-), dated December 15, 2014; for substitute approval of this change in type design applies to the above model aircraft only (2) Compatibility of this design change with previously approved modifications must be determined by the installer; (3) A copy of this Certificate must be maintained as part of the permanent records for the modified aircraft.

20/06/2018 Active 10056340

10056342 AIRGLAS, Inc. FAA STC SR00429AK R44 II R44 Installation of Airglas, Inc. LTB2600 Modular Terrain System.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.


Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

01/02/2016 Active 10056342

10056358 GULFSTREAM AEROSPACE CORPORATION FAA ST100279CD G200/GALAXY Installation of an Executive Cabin Interior of an Executive Cabin Interior.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

Flight Manual Supplement GA-21688M073H1<><> dated 11.Jul. 2003or later revisions of the above listed documents approved by EASA in accordance with EASA Dec Decision 2004/04/CF (or subsequent revisions offset this decision) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement Index:...

Limited to G200, S/N 073.... Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...

20/01/2016 Active 10056358


The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.

Master Document List, Ref. 21/073-C-OL-2000-001-03, dated 18/11/2016.Drawing Register, Ref. 21/073-C-DPF-2000-001-03, dated 18/11/2016; or later revisions of the above listed document(s) approved/accepted by the EASA system.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

15/06/2022 Active 10056375

10056381 HARTZELL PROPELLER INC. FAA STC SA03466CH PC-12-PC-12/45 PC-12/47 PC-12/47E Installation of Hartzell 5-blade composite propeller model H-ESA-341NC10245B.

The Certification Basis for the original product as amended/modified additional or alternative airworthiness requirements - Special Condition (s): See EASA approved Annex - Equivalent Safety Findings(s): See EASA approved Annex. The requirements for environmental protection and the associated/required change/repair it must be determined by the installer (2) Compatibility of this design change with previously approved modifications must be determined by the installer (3) A copy of this Certificate must be maintained as part of the permanent records for the modified aircraft.

-Jetpropeller Flight Manual Supplement, APS100-3014, dated April 1, 2014-Instructions for Installation, INST00314 dated May 23, 2015or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions offset this decision) and/or the Technical...

Propeller Limitations Diameter: 105 inches, no reductions permitted Pitch setting at 42 inch radius Lo: 14.7 +/- 0.1.0 degrees/Further: 50.0 +/- 0.5 degrees/Reverse: 17.5 +/- 0.5 degrees/Prior to installation of this design change it must be determined that the interrelationship between this design change...

21/01/2016 Active 10056381
**List of EASA Supplemental Type Certificates**

**10056384**

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<tr>
<td>LEARJET Inc.</td>
<td>REV. 1</td>
<td>FAA STD ST0173W</td>
<td>Installation of a 1194AC/60Hz/120VAC outlet in the right hand galley on the Learjet Model 45 aircraft.</td>
<td>1. Learjet Inc. Master Document List M100572ICT, revision (E), dated December 17, 2014 ((\lor)) FAA Approved December 28, 2014 or later FAA Approved revision 2. The following documents are required for MDL M100572ICT-01 configuration of this STC: Instructions for Continued Airworthiness M100572ICT-ICA-00.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. 2. Seven passenger floor plan (SA-3), install...</td>
<td>17/02/2016</td>
<td>Active</td>
<td>10058204</td>
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<th>STC Holder</th>
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<th>Certification basis</th>
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<tr>
<td>S4A SOLUTIONS</td>
<td>REV. 1</td>
<td>CESSNA DUESSLEDORF CITATIONSERVIC E CENTER GmbH</td>
<td>Dual Garmin GTN 750 or GTN750 and GTN650.Revision of the STCs to enable Iridium phone calls over the optional GMA35c audio panel. An Iridium Satphone must be already installed in the aircraft and is not included in this change.</td>
<td>- Engineering Order Doc. No. 2015-2-1. Rev. A. - Installation Procedure Doc. No. 2015-2-1. Rev. C. - Airplane Flight Manual Supplement Doc. No. 2015-2-10, Rev. C.</td>
<td>Vertical GPS/IBS based approach modes are not covered by this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/11/2019</td>
<td>Active</td>
<td>10058208</td>
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<td>DASSAULT FALCON SERVICE</td>
<td>REV.</td>
<td>DGAC FRANCE 2003-0209</td>
<td>MYSITRE-FALCON 900EX</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Applicability is limited to Falcon 900EX S/N 135F. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/01/2016</td>
<td>Active</td>
<td>10058234</td>
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**10056429**

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<th>STC Holder</th>
<th>Rev.</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>SAA SOLUTIONS OF AVIATION, S.L</td>
<td>REV.</td>
<td>737-400</td>
<td>TCAS V1.1 Upgrade</td>
<td>The Certification Basis for the original product is amended by the following or alternatively amended requirements: CS 25 Amendment 17CS 25.61&lt;(\lor)CS 25.27&lt;(\lor)CS 25.625&lt;(\lor)CS 25.25&lt;(\lor)CS 25.1301, CS 25.1302&lt;(\lor)CS 25.1309&lt;(\lor)CS 25.1317&lt;(\lor)</td>
<td>Applicable only to Boeing 737-400 aircraft with Honeywell TPA 81A TC/AC computer part number 035-50000-2220 previously installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/01/2016</td>
<td>Active</td>
<td>10058439</td>
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**10056456**

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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>GENEVA AVIATION Inc.</td>
<td>REV.</td>
<td>FAA STC S8270/5E</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Geneva Aviation Audio System on CS-29/Part 29 Category Rotorcraft in accordance with the enclosed EASA Approved Model List (AML). This certificate replaces the approval and the provisions of the EASA STC 10038866 for the CS-29/Part 29 rotorcraft identified in the enclosed AML.</td>
<td>A copy of this certificate, the Rotorcraft Flight Manual Supplement and/or Instructions for Continued Airworthiness must be maintained as part of the permanent records of the modified rotorcraft as listed in the AML of this approval. Prior installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change...</td>
<td>26/01/2016</td>
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<td>10058492</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>BAE SYSTEMS OPERATIONS LIMITED</td>
<td>REV. 1</td>
<td>HELICOPTER TAXI TECH, Inc.</td>
<td>Installation of a functioning Wireless In-Flight Entertainment System</td>
<td>BAE Systems Flight Manual Supplement BAES-STEM-44-001-90130-APM</td>
<td>1. Applicable to aircraft fitted with BAE Systems Minor ChangesSTMB0015A, STMB0016A and STMB0017A. Application of STC to other individual aircraft not fitted with BAE Systems Minor ChangesSTMB0015A, STMB0016A and STMB0017A must be by the user.</td>
<td>28/01/2016</td>
<td>Active</td>
<td>10056472</td>
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<tr>
<td>BELL</td>
<td>FAA STC SR09066RC</td>
<td>Installation of an Expanded Instrument Panel kit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Drawing List, Report No. AA-12013, Revision C, dated March 14, 2014.2. Instructions for Continued Airworthiness, Report No. AA-12008, NoRevision, dated October 10, 2013.</td>
<td>1. Eligible for installation on S/N 54300-Subsequent.2. The FAA has not certified the equipment for which these provisions are intended. Additional FAA approval is required for equipment installations.3. Prior to installation of this design change it must be determined that the interrelationship between...</td>
<td>29/01/2016</td>
<td>Active</td>
<td>10056476</td>
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<tr>
<td>BELL</td>
<td>FAA STC SR09065RC</td>
<td>Installation of an L-3 Wescam MX-10 System Kit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Rotorcraft Flight Manual Supplement, Report No. AA-12010, Revision C, dated May 9, 2014 or later FAA approved revisio2. Master Drawing List, Report No. AA-12004, Revision G, dated May 21, 2014.3. Instructions for Continued Airworthiness, Report No. AA-12009, NoRevision, dated October 8, 2015.</td>
<td>1. Eligible for installation on S/N 54300-Subsequent.2. High wind landing gear required for installation.3. Wescam MX-10 system GPS electrically disabled and not FAA certified.4. The FAA does not evaluate the Wescam MX-10 system/GPS antenna. The FAA must evaluate the Wescam MX-10...</td>
<td>29/01/2016</td>
<td>Active</td>
<td>10056478</td>
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<tr>
<td>HELI-ONE</td>
<td>TCCA STC SH07-24</td>
<td>Installation of Ryan 9900B8 Traffic Advisory System (TAS) in accordance with HelI-One Master Drawing List CHS76-34-040 MDL.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement FMS CHS76-34-040 Ryan 9900B8 Traffic Advisory System, Revision NC2. Master Drawing List, Report No. AA-12004-3, Revision G, dated May 21, 2014.3. Instructions for Continued Airworthiness, Report No. AA-12009, NoRevision, dated October 8, 2015.</td>
<td>Prior to the installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/01/2016</td>
<td>Active</td>
<td>10056483</td>
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<tr>
<td>HELI-ONE</td>
<td>TCCA STC SH06-13</td>
<td>Installation of Garmin GNS-530W GPS in accordance with HelI-One Master Drawing Lists CHS76-34-039 MDL (rotorcraft equipped with EFIS) and CHS76-34-039A MDL (rotorcraft without EFIS)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>For rotorcraft equipped with EFIS Flight Manual Supplement FMS CHS76-34-039C, Revision DFR rotorcraft without EFIS Flight Manual Supplement FMS CHS76-34-039A, Revision Dec later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU C...</td>
<td>The GNS-530W must have the WAAS function disabled. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>27/01/2016</td>
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<tr>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>C-212-DD, C-212-DF</td>
<td></td>
<td></td>
<td>TCAS V7.1 Upgrade</td>
<td>The Certification Basis for the original product as amended by later amendment(s) or additional or alternative airworthiness requirements. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>S4A Flight Manual Supplement A1205-HDK01 Issue 2 for later revisions of the above listed documents approved by EASA. The approval holder shall fulfill the obligations of Part 21, Chapter 21, Subchapter E, §§ 21.599-21.609. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change is maintained.</td>
<td>Applicable only to C-212-DD &amp; C-212-DF aircraft with Honeywell TPU 67A/68 A computer part number 098-01146-1211 previously installed. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change is maintained.</td>
<td>28/01/2016</td>
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<tr>
<td>ATELIER CHABORD</td>
<td>SCAC 45-SF-0001</td>
<td>REV. 3</td>
<td></td>
<td>Installation of exhaust system according to Chabord modification(STC-DR480-16-01 REV. 3, Enhanced silent block) - Alternative corrosion surface treatment - Enhanced material on different parts - Additional bracket on double rot tube - Enhanced shape of central structural part in muffler</td>
<td>The Certification Basis for the original product as amended by later amendment(s) or additional or alternative airworthiness requirements. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Technical Dossier ref. STC-DR485-16-01 Revision D dated 16 January 2022 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Chapter 21, Subchapter E, §§ 21.599-21.609. Prior to installation of this design change/repair it must be determined that the relationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/05/2022</td>
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<tr>
<td>INTER-TEC AERO DESIGN AND CERTIFICATION LTD</td>
<td>AB139, AW139</td>
<td>REV. 3</td>
<td></td>
<td>Modification Number ITS-139-25-013 - Forward Facing Observer Seat Rev. 1 - Extension of the applicability of the original STC to AB139/139/139 serial numbers and to introduce compatibility with Leonardo Helicopters floor panels 3G5332A05832 Rev. 2: Transfer of STC holder from INTER-TEC SERVICESA..</td>
<td>The Type and OXO Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Flight Manual Supplement ITS-033-016 Issue 1 - Instructions for Continued Airworthiness ITS-033-015 Issue 1 - Master Data List ITS-033-011 Issue 3 - Drawings ITS-033-007 Issue 4 Sheds 1 through 4 - Compliance Check List ITS-033-002 Issue 3 - Certification Plan ITS-033-001 Issue 40 revision.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>This STC is compatible only with Leonardo Helicopters floor panels 3G5332A05832. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>FAA STC</td>
<td>SA0765W-D</td>
<td></td>
<td>78</td>
<td>Installation of S-TEC System 60 Two Avra Automatic Flight Guidance System, Model ST-136, with Optional Automatic Electric Trim System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. FAA Approved Supplement to Pilot’s Operating Handbook and/or FAA Approved Airplane Flight Manual, PN 89127-1, dated 10.09.1996 and/or later FAA Approved revision of the above supplement. 2. Bulletin No. 209, Revision 2, dated 14.08.1988.3. Master Drawing List No. 52110, Revision A, dated 14.08.1988.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>SIRUM AEROTECH S.L</td>
<td>ATR72-212A</td>
<td></td>
<td></td>
<td>Flap Cam Guides substitution by TCCA PDA</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>1) TCCA “COMTEK Advances Structures Ltd.” Part Design Approval No. PG03.9 Issue 01, June 06, 2013. 2) Conversion Bulletin SA13-035-CB Rev. 0 Date: 04/12/2013 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canadair Ltd.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/02/2016</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
<th>StC Ref.</th>
<th>Description</th>
<th>Certification Basis</th>
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<tbody>
<tr>
<td>Gama Aviation (Engineering) Limited</td>
<td>REV. 3</td>
<td>CAT U. RESPONSIBIL ITY</td>
<td>B200T B200CT B200GT B200C B200</td>
<td>Approval of King Air 200/320 steep Approach capability up to 5.5 degrees. The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Change AD 1455-4 issue 4 to issue 6 dated 06 March 2018. Change AFMS-1455-01 issue 4 to issue 6-Instructions for Continued Airworthiness (ICA) in accordance with Gamaaviation Document ref. ICA-1455-01 Issue 2 dated 23 May 2016or later revisions of the above listed documents approved by EASA.</td>
<td>15/02/2018</td>
<td>Active</td>
<td>10056573</td>
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<tr>
<td>Scandinavian Avionics Design ApS</td>
<td>AS 332 L, AS 332 L1 AS 332 L2 AS 332 C, AS 332 C1</td>
<td>Single transponder system upgrade to MODE S EL/S EL/ADS-B</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Definition(s), CS 29.1309, CS ACNS.D.ELS.045, CS ACNS.D.AS332.155 as per CRI F-01 Issue 3. The following paragraph(s) at a later amendment: CS-ACNS Install/isue, Subpart D...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other pre-installed or other design change(s) or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/02/2016</td>
<td>Active</td>
<td>10056598</td>
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<tr>
<td>HaeCo Cabin Solutions, LLC</td>
<td>FAA STC ST0437AT</td>
<td>747-400</td>
<td>Installation of Collins SAT-906 SATCOM with Canadian-Maxarant OMA-2102Antenna System and Ball Airliner Low Gain Antenna. The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other pre-installed or other design change(s) or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/02/2016</td>
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<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>10056626</td>
<td></td>
<td>FAA STC</td>
<td>STO137W6-D</td>
<td>BD-700-1A10, BD-700-1A11</td>
<td>Installation of an Executive Interior Arrangement, including Cabin Seats, Jumpseats, Pocke Doo, Dvian, Passenger Oxygen System, Emergency Equipment, and Emergency Lighting System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>1. Jet Aviation Interim List 35-296G00, Rev. N, dated May 29, 2011 after FAA approved revision 2. FAA approved Airplane Flight Manual Supplement(s) 35-261M003, Rev. NC, dated March 27, 2006 (Config. 1), 35-261M005, Rev. NC, dated August 9, 2006 and (Optional) 35-261M006, Rev. NC, dated August 9, 2006...</td>
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<td>ARMSTRONG AEROSPACE Inc.dba ASTRONICS Aerospace</td>
<td>10056648</td>
<td></td>
<td>FAA STC</td>
<td>STO4034CH</td>
<td>A320-214</td>
<td>Installation of Passenger Cabin Seats and Inflight Entertainment (IFE)Upgrade.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>1. FAA Approved Flight Manual Supplement, Astronics Armstrong Aerospace Doc. No. DC303-6632-00, Revision -, approved September 03, 2015 or later FAA approved revision 2. Astronics Armstrong Aerospace Master Data List, Doc. No.DC104-6632-00, Revision A, dated September 2002, 2015...</td>
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<td>I.S.E.I INTERNATIONAL SERVEOSELECT RONIQUE INFORMATIQUE</td>
<td>10056667</td>
<td></td>
<td>APM 20, APM 30</td>
<td></td>
<td>Aerosol 2018 revision</td>
<td>Installation of SAFETYPLANE VS system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>None.</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10056671</td>
<td></td>
<td>FAA STC</td>
<td>ST0405H6AT-D</td>
<td>G280</td>
<td>Installation of an Automatic Dependent Surveillance Broadcast (ADS-B) Systemflight. 1 adds the Instructions for Continued Airworthiness.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any im...</td>
<td>- Gulfstream Index List GA411268000 Rev. *-, dated 10 April 2016- Flight Manual Supplement GA210- CGER-0030, Rev. *-, dated 10 April 2016- Instruction of Continued Airworthiness GA4113A4001 Revision. *-, dated 14 November 2015 later revisions of the above listed documents approved...</td>
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<td>TREB GmbH</td>
<td>10056719</td>
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<td>F260B</td>
<td></td>
<td>EO2020 Rev 2020</td>
<td>Fuel valve replacement</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>EOMD-2020 Rev 2020 – Dated 14.01.2016 – Engineering under Fuel systemModificationPUMD-0 20 Rev – Dated 14.01.2016 this is a supplement to Periodic Inspection List (Periodic Inspection list Jala Controls AIF Sial Marchetti F260-F260B – F260C – F260D Rev 1 “” referenced in TCDSDr later revisions of the above listed documents approved...</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>CRANFIELD AEROSPACE LTD</td>
<td>AVRO 146 SERIES RJ70</td>
<td>BAE 146-RJ70 ACTIVATION OF Wi-Fi SYSTEM</td>
<td>CRANFIELD AEROSPACE INSTRUCTIONS FOR CONTINUED AIRWORTHINESS SERVICEBULLETIN NO. CAADD/81428 ISSUE A-1</td>
<td></td>
<td>Restricted to ABN 12/64RRefer to CRANFIELD AEROSPACE TYPE DESIGN DEFINITION ADDENDUM NO.CADD/ DRO313 for the list the devices and quantities that may be usedwith this Wi-Fi network. Devices that are not in the list below must not be connected to the Wi-Fi network provided by the Rockwell CollinsWAP300...</td>
<td>12/02/2016</td>
<td>Active</td>
<td>10056720</td>
</tr>
<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>HAWKER 1000 BAE 125 SERIES 1000B &amp; BAE 125 SERIES 1000A</td>
<td>Indium Antenna Installation on B Ae 125-1000A, B Ae 125-1000B &amp; Hawker1000</td>
<td></td>
<td></td>
<td>Prior to installation of this design change it must be determined that...</td>
<td>12/02/2016</td>
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</tr>
<tr>
<td>ANORA INTERNATIONAL LLC</td>
<td>FAA STC ST0552LA</td>
<td>757-200 SERIES Installation of main deck cargo loading system.</td>
<td></td>
<td>FAA approved Ancra International LLC Controlled Data List DL85300,Revision D, dated February 24, 2015, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that...</td>
<td>15/02/2016</td>
<td>Active</td>
<td>10056734</td>
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<tr>
<td>ROEDER PRAEZISION GMBH</td>
<td>EMB-505</td>
<td>Installation of Low Gain Satcom Antenna</td>
<td>Master Document List No. LGA-NY-11277 Rev. 06 (Documentation ICA Instruction for Continued Airworthiness No LGA-NY-11277-07 Rev. 01or later revisions of the above listed documents approved by EASA.</td>
<td></td>
<td>Airworthiness limitations resulting from this modification according to...</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10056736</td>
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<tr>
<td>AERODATA AG</td>
<td>1900C</td>
<td>Installation of AD-AFIS-0113 in 1900C</td>
<td></td>
<td></td>
<td>Limited to model Beechcraft 1900C SN UC-10 and UC-33 only. The document JMM-1157 Rev. 1 (dated 10th February 2016) Chapter 04(discs) additional Airworthiness. Limitations applicable. Prior to installation of this design change it must be determined that...</td>
<td>15/02/2016</td>
<td>Active</td>
<td>10056738</td>
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<td>EASA STC No.</td>
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<td>Rev.</td>
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<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10056760</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCA STC SH13-22</td>
<td>117-C-1, 117-B-1, 117-B-2, 117-A-1, 117-A-3, 117-A-4</td>
<td>D117-837</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. Dart Aerospace Limited Flight Manual Supplement FMS- D117-837 Revision dated June 11, 2013 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU CAN/CA National Agreement-Master Document List MDL: D117-837 Revision J dated Feb...</td>
<td>Prior to installation of this design change it must be determined whether an interaction between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.</td>
<td>16/02/2016</td>
<td>Active</td>
</tr>
<tr>
<td>10056761</td>
<td>BOGLI AERONAUTICS S.R.L.</td>
<td>REV. 2</td>
<td>A8139, AW139</td>
<td>757-200</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Phoenix Aerospace Limited Flight Manual Supplement PAL/FMS/ A8139/106/Revision A dated 12 February 2016 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU CAN/CA National Agreement-Master Document List MDL: PAL/MDL/1297, Revision D dated 05 July 2016...</td>
<td>Applicable to rotorcraft without an ice protection system or limited ice protection system fitted. Prior to installation of this design change it must be determined whether an interaction between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.</td>
<td>05/03/2024</td>
</tr>
<tr>
<td>10056768</td>
<td>ICELANDAIR EHF</td>
<td>REV. 2</td>
<td>575-200</td>
<td>Installation of TCAS 7.1 Antenna Provision. Description of Type Design Change: Installation of an external doublet antenna adapter plate on the upper fuselage at STA 426.5 in accordance with Icelandic CIRA-757-5310-143/Revision J dated 1 June 2016. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Icelandic CIRA-757-5310-143/Revision J dated 1 June 2016...</td>
<td></td>
<td></td>
<td>02/06/2017</td>
</tr>
<tr>
<td>10056771</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 1</td>
<td>757-200</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>AES Master Data List AES-757-721MDL Issue 1...</td>
<td>Prior to installation of this change/repair it must be determined whether an interaction between this change/repair and any other previously installed change and/or repair will introduce an adverse effect upon the airworthiness of the product.</td>
<td>27/04/2021</td>
<td>Active</td>
</tr>
<tr>
<td>10056785</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>A320-214, A320-271N</td>
<td>Center Ceiling Stowage Installation. The change related to this STC deals with the installation of a Center Ceiling Stowage (CCS). The CCS is a small compartment unit located between C33 and C36 in the Ceiling and center area of the aircraft, used by the Cabin Crew, which means that the CCS is not...</td>
<td>Supplemental Type Definition Document CBD2015-13/35M Issue 1 rev 4-ICA, ref A-00-15-71-ICA, Issue 3 rev 1- Compliance Record List CRDL-15-25M (ax 1) and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.</td>
<td>This STC and the associated documents considered, are applicable only for A320-214 &lt;-&gt; A320-271N Models. The pre-requisite is installation of the above listed documents approved by EASA...</td>
<td>25/11/2016</td>
<td>Active</td>
</tr>
<tr>
<td>10056801</td>
<td>AVONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>C90A</td>
<td>LPV upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. Flight Manual Supplement ASR-2015-109-FMS-05-00 Rev. 08 for later revisions of the above listed documents approved by EASA.</td>
<td>ASR-2015-109-FMS-05-00 Rev. 08 for later revisions of the above listed documents approved by EASA...</td>
<td>A suitable HSI (e.g. Collins EHS-74) must have been already installed. Prior to installation of this design change it must be determined whether an interaction between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.</td>
<td>19/02/2016</td>
<td>Active</td>
</tr>
<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10056809</td>
<td>RAISBECK ENGINEERING Inc.</td>
<td>REV. 1</td>
<td>FAA STC SAUD735E</td>
<td>300, 300LW, B300, B300C</td>
<td>Installation of Raisbeck/Harbati Swept Blade Turboprop Propellers.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. FAA approved Airplane Flight Manual Supplement (AFM) listed installing Raisbeck Engineering Document 15-3505-DRL, Revision IR, dated January 25, 2016, or later FAA-approved revision. 2. Raisbeck Engineering Master Drawing List 15BC-A1-000, Revision IR, dated January 25, 2016, or later FAA-approved revision...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. This installation requires the incorporation...</td>
</tr>
<tr>
<td>10056823</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>TCCA STC SH12-28</td>
<td>AS 355F2, AS 355 N, AS 355 NP, AS 355 F1, AS 355 B, AS 355 B1, AS 355 B2, AS 350 B3, AS 350 BA, AS 350 D</td>
<td>AS350/AS355 Standard Interior modification including floor covering seat coverings, document holder and panel painting</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval: The requirements for environmental protection and the associated certifed noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>CP-0395 REV. H Standard Interior Modification AS350/AS355 ComplianceProgram MDL-11-030 REV. H Master Drawing ListCA-ECL-243 REV. 2 Instructions For Continued Airworthiness, seat coverCA-ECL-243 REV. 2 Instructions For Continued Airworthiness, floor coverCA-ECL-244 REV. 2 Instructions For Continued Airworthiness...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/02/2016</td>
</tr>
<tr>
<td>10056835</td>
<td>AIRBUS DEFENCE AND SPACE S.A.</td>
<td>A330-202-203-243</td>
<td>DCV0225S Modified Electrical Bonding of Cockpit Door</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airbus Defence and Space Declaration of Compliance Rev. TAE-FH-CO-150201 dated 16.11.2015 Certification Plan Rev. TAE-FH-DT-150033, A330 MRTT/FSTA Modified Electrical Bonding of Cockpit Door (DCV0225S) or later revisions of the above listed documents approved by EASA.</td>
<td>This STC can only be installed, if one of following EASA STC was previously installed: A330 MRTT EASA STC 10029972 or A330 MRTT STC 10034696 or A330 MRTT STC 10035495</td>
<td></td>
<td>22/02/2016</td>
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<tr>
<td>10056844</td>
<td>SABENA TECHNICS BGC</td>
<td>777F</td>
<td>777-300, 777-300ER 777-200, 777-200LR</td>
<td>Mechanical &amp; electrical installation for tablet computer/This STC deals with the installation on B777 of provisions both/mechanical (mounting assembly support on lower frame of eachcockpitbuilding window) and electrical (DC-DC USB converter/for personal tabletcomputer).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: All affected paragraph at CS25 amdt. 4The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval:...</td>
<td>- Master Data List 0688-01-A-MDL-RO-00-01 AFM Supplement 0688-01-A-01-01 AFMS-R01- Instructions for continued airworthiness 0688-01-A-ICA-R01 or later revisions of the above listed documents approved by EASA.</td>
<td>Only for Electronic Flight Bag use. No other use permitted. - In case of Electro Magnetic Interference when tablet computer is connected to electrical installation, the power supply cord must be disconnected from the USB outlet (power supply stops). It is the operator's responsibility to determine the...</td>
<td>23/02/2016</td>
</tr>
<tr>
<td>10056849</td>
<td>RSG PRODUCTS, Inc.</td>
<td>FAA STC SR001115DE</td>
<td>EC130B4, EC130T2</td>
<td>AS350D, AS350BA, AS350B1, AS350B2, AS350B3</td>
<td>Installation of a Video Camera System in accordance with Master Drawing List DL-4Note: This STC is the validation of FAA STC SR001115DE (date releasedAugust 15, 2014)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Integrated Flight Systems Inc. Master Drawing List DL-4 Revision FRSG Products Rotorcraft Flight Manual Supplement RFM-350-130- V5-01Revision B0r later revisions of the above listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/OF (or subsequent revisions oft this decision) and...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>EASA STC ref.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10056892</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td></td>
<td>A330-203</td>
<td>AERCAP Cabin rearrangement / PSS installation</td>
<td>The modification consists in cabin rearrangement from 243 passengers (30C + 30Y + 178 M) to 326 Economic and Rockwell Collins PW/ESTM/Passenger Service System (PSS) installation (no IFE).</td>
<td>Change Approval Sheet: CAS-A330-15-003 rev. 0. ICA - A330-15-003 rev 0 or later revisions of the above listed documents approved by EASA.</td>
<td>This STC is applicable to MSN 811, 819, 834 &amp; 900 only. Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce unadverse effect upon the airworthiness of the aircraft.</td>
<td>24/02/2016</td>
</tr>
<tr>
<td>10056904</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td>650</td>
<td>P-RNAV approval of existing Dual Honeywell GNS-XLS FMS System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Technical Order, doc. No. 2034649-70, dated 11 January 2016 - AERM Supplement, doc. No. 2034648-9, dated 16 February 2016.</td>
<td>None</td>
<td>26/02/2016</td>
</tr>
<tr>
<td>10056905</td>
<td>B/E AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST03890AT</td>
<td>737-600-700-800-900</td>
<td>Installation of a Collins SAT-2000 Satellite Communication System (SATCOM)-configuration 1 (B737-700/800).</td>
<td>Master Drawing List: 2-81151-001 rev FAM Supplement 2-81151-011 rev Altstructions for Continued Airworthiness, configuration 1 (B737-700/800) 2-81151-804 rev E-configuration 2 (B737-600-700-800-900) 2-13277-806 rev Acf later revisions of the above listed documents approved by EASA.</td>
<td>As per EASA the SATCOM communication system is not to be used during the approach, landing and takeoff phases of flight. Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce unadverse effect upon the airworthiness of the aircraft</td>
<td>26/02/2016</td>
</tr>
<tr>
<td>10056917</td>
<td>B/E AEROSPACE, Inc.</td>
<td></td>
<td>FAA STC ST02890AT</td>
<td>MD-80-30 MD-88 DC-8-61/62-61 DC-9-80 DC-9-15/-151/-21/-32/ F-135F/-34</td>
<td>Installation of a Kannad Model 406-AF Emergency Locator Transmitter (ELT) Revision 1 transferred from EMTEQ Engineering to B/E Aerospace.</td>
<td>Instructions for Continued Airworthiness (ICA): EMTEQ Engineering Document 2-23982-905, Rev. E, dated May 28, 2010.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce unadverse effect upon the airworthiness of the aircraft.</td>
<td>02/03/2016</td>
</tr>
<tr>
<td>(EASA.IM.A.S.00964)</td>
<td>ARMSTRONG AEROSPACE Inc dba ASTRONICS ARMSTRONG AEROSPACE</td>
<td></td>
<td>FAA STC ST02890CH</td>
<td>767-400ER SERIES 767-300 SERIES 767-200 SERIES</td>
<td>Certification of Compatibility with Cellular and Wireless Data Technologies.</td>
<td>Armstrong Aerospace Master Data List Doc. No. DC104-9414-00, Revision 1. Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce unadverse effect upon the airworthiness of the product. Limits for installation utilizing the...</td>
<td>29/02/2016</td>
<td>Active</td>
</tr>
<tr>
<td>10056934</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td></td>
<td>A330-243</td>
<td>Air Europa cabin rearrangement</td>
<td>This modification reconfigures the forward cabin area by main replace business Class seats Galley and In Flight Entertainment System.</td>
<td>Change Approval Sheet: CAS-A330-14-004 dated 24 February 2016 - Instruction for Continued Airworthiness document.</td>
<td>The EASA STC ref. EASA.A.S. 02890 Rev 1 (11001457 Rev. 1), or later approved revision, must be concurrently installed with this modification.</td>
<td>26/02/2016</td>
</tr>
<tr>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>STC Holder</td>
<td>INSTALLATION OF GOGO 2KU - DELTA</td>
<td>MOD 15179 - Glass Cockpit with G600 and MD302.</td>
<td>Airplane Flight Manual Supplement AE15179AFMS Rev. 0 dated 05 October 2015</td>
<td>FAASTARS Data List AE15179MLD Rev. 0 dated 10 December 2015 for later revisions of the above listed documents approved by EASA for Continued Airworthiness AE15179ICA Rev. 0 dated 05 October 2015.</td>
<td>Prior to installation of this change design it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/02/2016</td>
<td>Active</td>
<td>10056949</td>
</tr>
<tr>
<td>DELTA</td>
<td>REV. 3</td>
<td>FAA STC STO037NY</td>
<td>FAASTARS Data List (MDL) 0154-11329-2548 Revision dated 19 August 2021</td>
<td>Appendix 4 to Instruction Manual for Indoor Operations issue dated January 2011 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/12/2023</td>
<td>Active</td>
<td>10056957</td>
</tr>
<tr>
<td>BALLONBAU</td>
<td>Indoor operation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/this approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Universal FMS UMS-1LW installation</td>
<td>The aircraft must have other previously installed design change/repair and any interrelationship between this design change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>The STC is applicable when the tethered balloons are operated indoors with a roof. During these operations an additional crew member must be available to help the ground operator.</td>
<td>01/03/2016</td>
<td>Active</td>
<td>10056959</td>
</tr>
<tr>
<td>SCANDINAVIAN</td>
<td>AS 332 L2</td>
<td>Installation of Skywatch TRC899 including antenna, GTN625 including antenna, Flightstream210, Indium antenna and USB charger.</td>
<td>Flight Manual Supplement 2097844-AM Issue 4 dated 16.02.2016</td>
<td>- The aircraft must have other previously installed design change/repair and any interrelationship between this design change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>The STC is applicable when the tethered balloons are operated indoors with a roof. During these operations an additional crew member must be available to help the ground operator.</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10057013</td>
</tr>
<tr>
<td>RHEINLAND AIR</td>
<td>P. 180 AVANTI</td>
<td>Installation of Skywatch TRC899 including antenna, GTN625 including antenna, Flightstream210, Indium antenna and USB charger.</td>
<td>Flight Manual Supplement 2097844-DCC Issue 2 dated 16.02.2016</td>
<td>- The aircraft must have other previously installed design change/repair and any interrelationship between this design change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>The STC is applicable when the tethered balloons are operated indoors with a roof. During these operations an additional crew member must be available to help the ground operator.</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10057014</td>
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<tr>
<td>STC Hold</td>
<td>Rev.</td>
<td>Foreign/NAAA Chn</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>STC HOLDING COMPANY, INC.</td>
<td>REV. 1</td>
<td>FAA STC SA6215SW</td>
<td>DHC-6</td>
<td>Installation of modified passenger door.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/06/2021</td>
<td>Active</td>
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<tr>
<td>NORTHERN AVIONICS S.R.L.</td>
<td></td>
<td>A110, AY119MKII</td>
<td>First Limit Advisory System (FLAS)/The FLAS system is designed to trigger by means of a discrete output the Rotorcraft Flight Manual (including GSM personal data and Low Visibility Approach Operation, use &quot;YFR On&quot;/&quot;OFF&quot; function located on the CCP on the top of FWD Li Attendant's Area above attendant seat TEDPs (including GSM personal devices) are prohibited to work and...</td>
<td>02/02/2016</td>
<td>Active</td>
<td>10057020</td>
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<tr>
<td>AEROTECH &amp; CONCEPT</td>
<td></td>
<td>767-300</td>
<td>Airborne wireless platform activation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The FLAS installation is applicable only on R44 II and R44 II Helicopter installing 1- Electrical Display Unit Installation Document (RP) DPN191-3103 or P/N 200/11-31034-200/11-31034-21</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10057017</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>FAA SR02254LA</td>
<td>R 44, R44 II</td>
<td>Installation of stability augmentation system and autopilot/inconformance with S-TEC Master Data List, HelisAS-026. This STC is the validation of FAA STC SR02254LA (date amended/November 12, 2014).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>As per AFMS, WiFi system is prohibited to work and must be switched OFF during P&amp;P and Low Visibility Approach Operation, use &quot;YFR On&quot;/&quot;OFF&quot; function located...</td>
<td>02/03/2016</td>
<td>Active</td>
<td>10057027</td>
</tr>
<tr>
<td>FALCON CREST AVIATION SUPPLY, INC.</td>
<td>REV. 1</td>
<td>FAA SR00504RC</td>
<td>206A, 206 B</td>
<td>Replacement of the original NiCad Battery with the ConcordeRG-206 valve-regulated sealed acid battery.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/03/2016</td>
<td>Active</td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>STC Type</td>
<td>Rev.</td>
<td>Foreign/NAA STC No.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10057046</td>
<td>AVCON INDUSTRIES, Inc.</td>
<td>FAA STC</td>
<td>1</td>
<td>SA03598AT</td>
<td>B300, B300C, B200GT, B200COT 260, 200C, B200, B200C</td>
<td>Installation of a Supplemental DC Power for an Isolated Instrumental Bus.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1) Aviation Design Services Master Drawing List, ADS R801, Revision B dated May 22 (rev. G) 2009 (rev. G) Avcon Industries, Inc., Master Data List, ADS R601, Revision C, dated January 15 (rev. 0) 2012, or later FAA approved revisions.</td>
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<td>10057051</td>
<td>AMTECH AERONAUTICAL LIMITED</td>
<td>TCCA STC</td>
<td>1</td>
<td>NO. SH02-40</td>
<td>AS 350 B, AS 350 B1, AS 350 B2, AS 350 B3, AS 350 BA, AS 350 D</td>
<td>Spectrolab &amp; Trakkabeam Searchlight Installation. This STC is the validation of Canadian STC No. SH02-40.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation of Spectrolab SX-5/SX-16XP Engineering Order EO342401, Revision 5, dated 17 March 2011 - Flight Manual Supplement FNE0942401, Revision 4, dated 17 August 2015 - Maintenance Manual Supplement FNE0842401, Revision 5, dated 8 March 2011 Installation of Trakkabeam A500V400 (-1)(A)00001K8 - Engine...</td>
</tr>
<tr>
<td>10057054</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>TCCA STC</td>
<td>1</td>
<td>SH08-43</td>
<td>AS 355 F2, AS 355 N, AS 355 MP, AS 355 E, AS 355 F1, AS 355 F, AS 355 B1, AS 355 B3, AS 355 B2, AS 355 B3, AS 355 BA, AS 355 D</td>
<td>Installation of Rear Crosstube Maintenance Steps (350-20000). This STC is a validation of Canadian STC No. SH08-43.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Airbus Helicopters Canada Limited Master Drawing List MDL-98-002, Issue C, Transport Canada approved 06 December 2014 - Airbus Helicopters Canada Limited Instructions for continued Airworthiness ICA-ECL-235, revision 1, Transport Canada accepted 09 December, 2014 for later revisions of the above listed douc...</td>
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<td>10057056</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC</td>
<td>1</td>
<td>ST00845SE</td>
<td>737-700, 800, 900, -900ER</td>
<td>Installation of the Head-Up Display Guidance System HS5 Model 4000</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI A-01 Issue 2 &quot;Certification Basis&quot; (Rev. 5) as applicable. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change(s) and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>- FAA Approved Airplane Flight Manual Supplement for the Boeing 737-700, 700C, 700, 900, and -900ER Series Model 4000 Head-Up Guidance System (HS5) Ref. 523-0809596, Rev. 5. FAA Approved Airplane Flight Manual Supplement for the Boeing 737-800 Series Model 4000 Head-Up Guidance System (HS5) Ref. 523-0809568.</td>
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<td>10057058</td>
<td>COMTECH GLOBAL AVIATION SERVICES S.</td>
<td>REV. 1</td>
<td>A320-214</td>
<td>A319-115, A319-112</td>
<td>Introduction of Structural Provisions for Cabin Access Points</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI A-01 Issue 2 &quot;Certification Basis&quot; (Rev. 5) as applicable. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change(s) and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>CTE-M80141-320 Modification Summary - CTE-W80141-01 Weight &amp; Balance Report - CTE- EO0414-01 Engineering Order, A319 aircraft - CTE- EO0414-02 Engineering Order, A320 aircraft - CTE- EO141-01-320 Structural Provision Installation later revisions of the above listed document(s) approved acceptable thru...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change(s) and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
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<td>Type Certificate Number</td>
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<td>Certification Basis</td>
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<td>1005708</td>
<td>VT MOBILE AEROSPACE ENGINEERING, Inc</td>
<td>10057076</td>
<td>Modification of Passenger Aeroplane to 15-Pallet Cargo ConfigurationRevision 1: Limitation related to winglets revised</td>
<td>The Certification Basis for the original product as amendability to installing additional or alternative airworthiness requirements JAR-25 Change 11: 25.812a(3)(x)(y)(z)(A)(R)(H)(I) ([JAR-25 Change 13: 25.321, 25.331, 25.335, 25.341, 25.345, 25.349,25.351, 25.361(1)(f)(g), 25.365(f)(1)(c), 25.37…])</td>
<td>It has been demonstrated that this modification is compatible with Boeing Model 757-200 Aeroplanes with winglets installed per Aviation Partners Boeing STC ST010185E(-)(-). &gt; validated by EASA STC 1001569964. No other winglet installations have been evaluated. This modification must be installed concurrently.</td>
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<tr>
<td>1005706</td>
<td>HELI-ONE (NORWAY) AS</td>
<td>AS 332 L1</td>
<td>Installation of pitch locks</td>
<td>The Certification Basis for the original product as amendability to installing additional or alternative airworthiness requirements Equivalent Safety Findings (c) CS29.811(i) Photo luminous exit identifiers recorded in CRI D-010 following paragraph(s) at a later amendment. Elected to comply with CS-2.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10057091</td>
<td>WIPARE Inc.</td>
<td>FAA SP03786CH</td>
<td>Installation of pitch locks</td>
<td>Wipaire, Inc. Installation Certification No. 10005830, Revision A, dated August 8, 2014.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10057098</td>
<td>WIPARE Inc.</td>
<td>FAA SA03791CH</td>
<td>Installation of McCasley Model 3GFR34703 propeller, modified to include pitch locks per STC SP03786CH.</td>
<td>Wipaire, Inc. Drawing List AHUK/1353030-001, Issue 0 or later FAA approved revisions.2. Airplane Flight Manual Supplement, Document No. W208PL-25-01, Revision, dated August 27, 2016.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<td>10057114</td>
<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>REV. 3</td>
<td>Installation of a Rockwell Collins Pro Line 21 update to and including TAWS mod-5 warnings during LPV approaches, optional AOA-B Outfit and optional synthetic vision system (SVS) Rev. 3 adds serial numbers of Textron Aviation B200 and B300 airframe and Pro Line 21 displays.</td>
<td>1. Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10057145</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>FAA STC</td>
<td>Installation of Trakka A800 Search Light on the forward end of the left hand landing gear step (AHUK/1353030) in accordance with Drawing List AHUK/1353030-001.</td>
<td>Flight Manual Supplement Installation of Trakka A800 Search Light AHUK/1353030-001, Issue B.</td>
<td>As prerequisite L1 Mewker Aviation Ap fiancy Step Utility Mount in accordance with the FAA STC 10017058 must be installed. Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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**List of EASA Supplemental Type Certificates**

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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>System Record</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST0405AT-D</td>
<td>G280</td>
<td>Activation of JUNK 2000+ software function.</td>
<td>The Certification Basis for the original product as amended to include additional or alternative airworthiness requirements. Special Condition (s)/CRIF-F-38 Data Link Services for Single European SkyCRIF F-37 Flight Recorders including Data Link Recording. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended and remain applicable to this certificate/approval.</td>
<td>- Aircraft Avionics Suite 3/4 and later EASA approvals, required for this STC. Following STC is required to be installed for this STC: 10052489. Installation of an Executive Cabin Interior. Prior to installation of this modification, it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a adverse effect upon the airworthiness of the product, the modified system must meet...</td>
<td>09/03/2016</td>
<td>10052489</td>
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<tr>
<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>FAA ST0325LA</td>
<td>400A</td>
<td>Installation of Rockwell Collins Flight Management System FMC 5000 software Version 4.0.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended and remain applicable to this certificate/approval.</td>
<td>1. FAA approved Airplane Flight Manual Supplement, C600/04/04, revision 0, dated 06 December 2010 or later. FAA approved revision 0.</td>
<td>14/03/2016</td>
<td>10057180</td>
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<tr>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>FAA SR007135E</td>
<td>REV. 4</td>
<td>Installation of Onboard Systems International Cargo Hook Suspension Kit in accordance with master drawing list 155-048-00.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended and remain applicable to this certificate/approval.</td>
<td>- Rotorcraft Flight Manual Supplement Cargo Hook Suspension Kit Document Number 121-001-00 Revision 4, dated August 1, 2012. - Rotorcraft Flight Manual Supplement Cargo Hook Suspension Kit Document Number 121-000-00 Revision 0, dated October 13 2011 for later revisions of the above listed documents applicable to this certificate/approval.</td>
<td>Approval of this change in type design is only applicable to the models listed in this certificate which were previously equipped with Bell Cargo hook suspension assembly Part Number 204-072-915-25 or Part Number 204-072-915-103. The Cargo Hook suspension systems, as installed in accordance with theapp...</td>
<td>13/01/2017</td>
<td>10057190</td>
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<tr>
<td>DRF STIFTUNG LUFTRETTUNG</td>
<td>EC135 P2/ EC135 P2+</td>
<td>REV. 2</td>
<td>Installation of GNC 256B and GTC 650 COM/NAV System.</td>
<td>The Certification Basis for the original product as amended to include additional or alternative airworthiness requirements. The following paragraphs (s)/CRIF F-37 are applicable to the above listed documents. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement FMS 10-DRF-033, Revision 05, Supplement for G/A 256B dated January 14, 2016 or later. Prior to installation of this STC: 10052498, Installation of Link 2000+ software. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a adverse effect upon the airworthiness of the product.</td>
<td>Limited to VFR operation only. Prior to installation of this design change it must be determined that the design change and any other previously installed design change and/or repair will not introduce a adverse effect upon the airworthiness of the product.</td>
<td>08/04/2016</td>
<td>10057220</td>
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<tr>
<td>AVIANOR INC</td>
<td>A330-243</td>
<td>REV. 1</td>
<td>Modification of Recaro Model 6610 passenger seats by changing dresscovers (r/s) holding time and installing Zi RAVE IFE equipment.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended and remain applicable to this certificate/approval.</td>
<td>AVIANOR MASTER DOCUMENT LIST MDL.1412601 REV. NC DATED 15 DECEMBER 2015</td>
<td>NOTE THAT THIS APPROVAL IS FOR SEAT MODIFICATIONS ONLY. INSTALLATION OF THE MODIFIED SEAT ONTO THE AIRCRAFT AND THE RAVE IFE EQUIPMENT IS TO BE IN ACCORDANCE WITH EASA STC REFERENCING TO EASA PROJECT NO. 001004566R Restricted to ASINs 261 and 265.</td>
<td>04/04/2016</td>
<td>10057244</td>
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| 1005726 | BOMBARDIER Inc. | TCGA LSA13-07/1D | BD-700-1A10 | Display of EVS (Enhanced Video System) Video on Copilot’s MFD | The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval. | Modification in accordance with ACS-MAI Ltd., Modification Data Summary J13300 Revision A:Refer to Maintenance Manual Supplement J13301 Revision NC or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement. Supplemental Type Certificate limited to serial number 9264
| 1005724 | DUNLOP AIRCRAFT TYRES LIMITED | CAA UK RESPONSIBILITY | DHC-8-315 DHC-8-314 DHC-8-311 DHC-8-301 | Installation of Duniop P/N DR 26222T Bias Main Landing Gear Tyre(s) on the DHC-8-300 Series | The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: CS25.105(a)(c), 25.109, 25.125(a)(b)(e)(f), 25.233 (a)–(c), 25.235, 25.305, 25.471, 25.473, 25.477, 25.481, 25.483. | Installation has to be in accordance with DATL - PR11 - MD in later revisions of the above listed documents approved by EASA/DAE - PR01 - ICA - Instructions for Continued Airworthiness/DATL - PR01 - AMMS - Aircraft Maintenance Manual
| 1005728 | AMAC AEROSPACE SWITZERLAND AG | REV. 3 | 737-800 | Replacement of Large Monitors, Replacement of the currently installed large monitors with new larger monitors of less weight as defined in the Master Data List, Doc. No. MDL-AR10-4420, Rev. B, dated 16.03.2016, | The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: Special Condition (a): CRI D-10, Glass Screens of Displays / Monitors. The requirements for environmental protection and the associated certification noise and/or emissions are... | Master Data List, Doc. No. MDL-AR10-4420, Rev. B, dated 16.03.2016
| 1005719 | AVIATION TRADERS LIMITED EUROPE S.A.S. | REV. 1 | 787-300ER | Aviation Traders Limited Modification ref. MOD-11930-0001 issue 1. Installation of Crew Rest Module. This Modification installs a pallet-mounted crew rest module of similar dimensions as standard A2 Unit Load Device in the A1 cargo container position in front of the main cargo deck in accordance with the applicable to this certificate. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25.105(a)(c), 25.109, 25.125(a)(b)(e)(f), 25.233 (a)–(c), 25.235, 25.305, 25.471, 25.473, 25.477, 25.481, 25.483. | ATL MDL ref. MDL-11930-0001 issue 7 - ATL AFS ref. 11930- AFS-0001 issue 1 - ATL Modification Summary Sheet MOD-11930-0001 issue 10 - later revisions of the above listed document(s) approved by acceptance/under the EASA system.
| 1005732 | COMTECH GLOBAL AVIATION SERVICES S. | REV. 1 | A319-111 | Introduction of 144 passenger interior configuration and associated cabin changes | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25 Amendment 17 (various). The requirements for environmental protection and the associated certification noise and/or emissions are... | COMTECH Aviation Services Master Data List No. CTE-MDLU142-319 Rev. ...dated 18 March 2016 for later revisions of the above listed document(s) approved/acceptance/under the EASA system.
| 1005734 | JCB AERO 9A | Revision 18 | 757-200 | Boeing 757 Cabin Modification B752-24527-03 initial issue. The modification consists in replacing the In-Flight Entertainment systems, modifying the lighting systems, installing a new In-Flight Power Supply System and re-arranging the monuments on-board a VIP configured Boeing B757-200 in accordance with JCB Aeronautical Services, 17-020. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (a)–(b) 01 Iss.01 & 02 Iss. 01 The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product... | JCB Aeronautical services/Boeing 757-24527-03 initial issue/01 Iss. 01 The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product... | Limited to MSN 0045726 only/ACP-002 (s) converted by this STC must be (a) operated in accordance with the latest specification of the Boeing 757-24527-03 aircraft. This STC must be (a) operated in accordance with the latest specification of the Boeing 757-24527-03 aircraft. This STC must be (a) operated in accordance with the latest specification of the Boeing 757-24527-03 aircraft. This STC must be operated in accordance with the following JCB AERO document...
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<tr>
<th>EASA STC No.</th>
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<th>Foreign/NAACSTC ref.</th>
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10057327</td>
<td>JCB AERO SA</td>
<td></td>
<td>757-200</td>
<td></td>
<td>B757 VIP lavatory installation Modification No. B775-24527-07 InitialRelease. The modification consists in replacing the thermostat with a new V/Plenum between Door 1 and Fuselage Station 461, in accordance with JCB Aero Master Drawing List reference: A105-24527-07-MDL-01 Revision A</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>- JCB AERO ICAO reference B752-24527-07-ICA-01 Initial Release</td>
<td>Later revisions of the above listed documents approved by EASA - JCB AERO Service Bulletin ref. B752-24527-07, SB01Revision A, JCB AERO Master Drawing List ref. B752-24527-07-MDL-01Revision A.</td>
<td>Limited to MSN 24527 only.Aeronet(s) converted by this STC must be maintained in accordance with the following JCB AERO document: Supplemental instructions for Continued Airworthiness, Document B752-24527-07, ICA-01 Initial Release.JCB AERO Modification B752-24527-07 is a concurrent modification to:</td>
<td>22/02/2016</td>
<td>Active</td>
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<td>10057328</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td></td>
<td>A319-112</td>
<td></td>
<td>Airbus A319 Cabin WiFi System Installation for IFE streaming</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>- H4 Aerospace Flight Installation Manual Supplement No. H4AFMS040 Revision Initial-H4 Aerospace Instructions for Continued Airworthiness No. H4ADR072Issue 2or later revisions of the above listed documents approved by EASA - H4 Aerospace Modification Summary No. H4A1910 Issue 1 dated 18 March 2016</td>
<td>Restricted to MSN’s 1598 1625Future modifications must consider the WiFi 2.4GHz and 5GHz RFenvironment during the design.</td>
<td>22/02/2016</td>
<td>Active</td>
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<tr>
<td>10057329</td>
<td>CRANFIELD AEROSPACE LTD</td>
<td></td>
<td>JETSTREAM SERIES 3100</td>
<td></td>
<td>Installation of a Flight Navigation System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25 Amendment 425.603; 25.605; 25.611; 25.613; 25.771; 25.773; 25.789; 25.899; 25.1302; 25.1309; 25.1310A; 25.1325The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>Cranfield Supplement No. 5 Issue 3 dated July 2014 to Jetstream CAA Approved Flight manual Doc. No. HP. 41.15or later revisions of the above listed documents approved by EASA-Cranfield Aerospace Ltd Type Design Definition Addendum CADD9813 Issue 6 dated 21 March 2016Cranfield Aerospace Ltd Declaration of Conformity</td>
<td>Aircraft Serial Number 614 only. Prior to installation of this design change it must be determined thatthe interrelationship between this design change and any otherpreviously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>22/02/2016</td>
<td>Active</td>
<td>10057329</td>
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<tr>
<td>10057330</td>
<td>TASS-EU Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>JETSTREAM SERIES 3100</td>
<td>Installation of GPS Antenna provisions</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25 Amendment 17; 25.301; 25.303; 25.305; 25.307; 25.601; 25.603; 25.605; 25.607; 25.609; 25.611; 25.613; 25.625; 25.1310A; 25.1325The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>TASS EU Change Classification and Certification Sheets No.s 101048-CCSa-132 Revision 10 dated 21 March 2017 and 101048-CCSa-132Revision 10 dated 21 March 2017</td>
<td>None</td>
<td>21/03/2017</td>
<td>Active</td>
<td>10057330</td>
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<td>10057351</td>
<td>VT DRB AVIATION CONSULTANTS, INC.</td>
<td>REV. 2</td>
<td>FAA STC ST11167SC</td>
<td></td>
<td>VIP Interior Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s) CRI-D-02 Isolated Completely CRI-D-05 Glass Screens of Displays and Monitors CRI-D-06 Shower Installation CRI-D-08 Injury Criteria for Side Facing Diva...</td>
<td>VT DRB AIRPLANE FLIGHT MANUAL SUPPLEMENT No. 8052CT25-01 Revision A VT DRB Instructions for Continued Airworthiness No.8052CT25-03 Revision or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions offthis decision) and...</td>
<td>Restricted to MSN 60406EASA STC No.s 10053109 &quot;Structural modification necessary for the installation of an EMS ALT-350 SATCOM antenna (&lt;(&gt;) &amp;&lt;&gt;) EASA-IM.A.S. 5308/Installation of All Wing Failing Acoustic Thermal Installation Kit (&lt;(&gt;) &amp;&lt;&gt;) and...10056122 &quot;Installation of an Auxiliary Fuel System&quot; must be in...</td>
<td>14/04/2016</td>
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<td>Limitations/Conditions</td>
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<td>Associated Technical</td>
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### List of EASA Supplemental Type Certificates

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<th>Registration Code</th>
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<th>Type Certification</th>
<th>Airworthiness Certification</th>
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<tr>
<td>AEROTEC &amp; BLACKHAWK</td>
<td>EC 130 T2</td>
<td>Provision for ENGINE AIR INTAKE COVER installation, P/N 350-700924.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>31/03/2016</td>
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<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA STC SA5246LA</td>
<td>Installation of Pro &amp; v/w PTPSA-140 and 3-blade Hartzell Propeller/HC-657N-SAF (HYT89806BNS) 2-70.</td>
<td>FAA approved Master Drawing List 2014-01000, Rev. IR, dated April 7, 2015.</td>
<td>05/04/2016</td>
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<td>BOND HELICOPTERS EUROPE Ltd.</td>
<td>EC135 T2+</td>
<td>Norwegian Aircraft Reconfiguration – BOND/MOD/118 - NIVS – EC135Description: This change fits two Master steps to carry an MX-15 camera and a flightDeck AP searchlight. The aircraft is compatible with Night Vision Goggles (NVG).</td>
<td>Aircraft Flight Manual Supplement ATG-MV-ECJIL135-02*01, dated 09/2015.</td>
<td>18/04/2016</td>
<td>Active</td>
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The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Appendix 1. Certification Basis (1) The type certification basis for Model 412EP helicopter as shown in EASA, IM: R-108 remains valid for 412EP parts not impacted by this change...

1) This STC is limited to 412EP aircraft serial number 36248 and subsequent or those otherwise modified to be equivalent with regards to AHRS (LIR-100) wiring and installation. For aircraft serial numbers 36248 through 36490, the dual LCR-92 AHRS installation must be replaced with the dual LCR-100.

10057662  EMBRAER GPX Ltda (PC 725) ANAC BRAZIL REF 2013S09-09 EMB-505 Installation of Pilot’s new extended seat tracks.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce a adverse effect upon the airworthiness of the product.

10057665  DAVIS AVIATION SERVICES FAA STC SA2830SO U206, U206A THROUGH G 206, P206, P206A THROUGH E Installation of a Telakomye Continental Motors (TOM) model IO-550-E5B50B engine, with either a McCauley model D3A34C200G0FA10-D1, D3A34C4011G0DF0A10, D3A34C4011G0DF0A10, or a Harriell model HC-23YV-1RF, FL87C13H9, PHC-23YV-1RF, FA-8468A1-9R, PHC-23YF-1RF, FA-8468A8-9R, PHC-23YF-1RF.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce a adverse effect upon the airworthiness of the product.

10057671  HARTZELL PROPELLER INC. REV. 1 FAA STC SR03474CH Installation of Heli-Preheat Kit, p/n TA3083-1 for 115 volt preheating installation of Heli-Preheat Kit, p/n TA3083-2 for 230 volt application.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.


Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce a adverse effect upon the airworthiness of the product.

10057674  HARTZELL PROPELLER LLC REV. 1 FAA STC SR03168CH Installation of Heli-Preheat Kit, p/n TA5854-1 for 115 volt preheating installation of Heli-Preheat Kit, p/n TA5854-2 for 230 volt application.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce a adverse effect upon the airworthiness of the product.

10057681  EMBRAER GXP Ltda (PC 725) ANAC BRAZIL REF 2013S09-09 EMB-505 Installation of Pilot’s new extended seat tracks.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.

10057682  EMBRAER GXP Ltda (PC 725) ANAC BRAZIL REF 2013S09-09 EMB-505 Installation of Pilot’s new extended seat tracks.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.

10057683  DAVIS AVIATION SERVICES FAA STC SA2830SO U206, U206A THROUGH G 206, P206, P206A THROUGH E Installation of a Telakomye Continental Motors (TOM) model IO-550-E5B50B engine, with either a McCauley model D3A34C200G0FA10-D1, D3A34C4011G0DF0A10, D3A34C4011G0DF0A10, or a Harriell model HC-23YV-1RF, FL87C13H9, PHC-23YV-1RF, FA-8468A1-9R, PHC-23YF-1RF, FA-8468A8-9R, PHC-23YF-1RF.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.

10057685  DAVIS AVIATION SERVICES FAA STC SA2830SO U206, U206A THROUGH G 206, P206, P206A THROUGH E Installation of a Telakomye Continental Motors (TOM) model IO-550-E5B50B engine, with either a McCauley model D3A34C200G0FA10-D1, D3A34C4011G0DF0A10, D3A34C4011G0DF0A10, or a Harriell model HC-23YV-1RF, FL87C13H9, PHC-23YV-1RF, FA-8468A1-9R, PHC-23YF-1RF, FA-8468A8-9R, PHC-23YF-1RF.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.

10057691  HARTZELL PROPELLER INC. REV. 1 FAA STC SR03474CH Installation of Heli-Preheat Kit, p/n TA3083-1 for 115 volt preheating installation of Heli-Preheat Kit, p/n TA3083-2 for 230 volt application.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.

10057692  EMBRAER GXP Ltda (PC 725) ANAC BRAZIL REF 2013S09-09 EMB-505 Installation of Pilot’s new extended seat tracks.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.

10057693  HARTZELL PROPELLER LLC REV. 1 FAA STC SR03168CH Installation of Heli-Preheat Kit, p/n TA5854-1 for 115 volt preheating installation of Heli-Preheat Kit, p/n TA5854-2 for 230 volt application.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.

10057694  HARTZELL PROPELLER LLC REV. 1 FAA STC SR03168CH Installation of Heli-Preheat Kit, p/n TA5854-1 for 115 volt preheating installation of Heli-Preheat Kit, p/n TA5854-2 for 230 volt application.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this change, it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.
10057677 EICHELBERGER FLUGZEUGBAU STC Holder: DULO DISCUS Modification of the fuselage for Installation of a jet-engine The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.
- Modification of the fuselage for Installation of a jet-engine (Aerdynamics, Rumples zum Einbau eines Jet-Train/works) refer to the above listed documents approved by EASA. Project description: List of modification-Load Test-Restoration of fuselage and tempering-Pictures of the de-instal...
- Flight Operation manual Updates for Duo Discus SN 204/Prior to installation of this design change it must be determined that the relationship between this design change and any theretofore installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product...

03/04/2016 Active 10557677

10057679 BHE & ASSOCIATES, LTD. FAA STC: SA11347C Installation of a Rockwell Collins Pro Line 21 update to include TAWS/Smoke 5 warnings during LPV approaches, ADS-B Out and optional synthetic vision system (SVS).
New Certification Basis: Based on 14 CFR §§ 21.115 and 21.101, and the FAA policy for significant changes in FAA Order 8110.48, the certification basis for the Beechcraft Corporation/8300 and B300C modified by installing the Rockwell Collins Pro Line 21 update to include TAWS/Smoke 5 warnings during LPV approach.

08/04/2016 Active 10557679

10057682 GYRON SYSTEMS INTERNATIONAL, Ltd. FAA STC: SR01235A Installation of a Nose Window Utility Mount in accordance with FAA approved Gyron Systems International, LTD Master Drawing List AFNW-010.
The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.

06/04/2016 Active 10557682

10057686 STG AEROSPACE GERMANY GmbH REV. 2 DHC-8-402 DHC-8-402L1 DHC-8-402L1 DHC-8-402L1 DHC-8-402L1 DHC-8-402L1 DHC-8-402L1 PL Aisle Path Marking SYSTEM:Revision 01 SuperSeal Ultralight (SSUL) Blue emiting and additional heat inserts amdeals/heat resistant additions.Added Revision 2. Transfer of STC from STG Aerospace to STG Aerospace GmbH.
The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment for Airbus DHC-8-402, DHC-8-401, DHC-8-401,

12/02/2016 Active 10557686

10057696 CONCORDE BATTERY CORPORATION FAA STC: SA16953A Installation of Concorde Battery RG-380E/40L.
The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.
- 1. Concorde Battery Corp FAA Approved Airplane Flight Manual Supplement No. 5-2239, FAA approved on September 10, 2004 or later FAA approved revisions 2. FAA approved Concorde Battery Corporation Master Drawing List Drawing No. 5-0127. Revision "B", dated August 20, 2004, or later FAA-app...

07/04/2016 Active 10557696
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<th>Certification basis</th>
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<tr>
<td>10057702</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>BELL 412/412EP</td>
<td>Installation of Sandel Avionics ST3400H HTAWS</td>
<td>The Certification Basis for the original product as amended by the following: additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-29 Amdt 3 paragraph CS29(1), CS 29.25, CS 29.301, CS 29.303, CS 29.307, CS 29.321, CS 29.341, CS 29.601, CS2...</td>
<td>- OEM Supplement Doc. 2085628-AM Issue 6 dated 22.02.2016;</td>
<td>Prior to installation of this design change must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/04/2016</td>
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<td>10057703</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES NC</td>
<td>A330-203</td>
<td>Economy Class Seat Replacement of EVA Air Aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Summary, Northwest Aerospace Technologies, Inc. document No. N484-D00, Revision A, dated 03/07/2024; Compliance Record to MG 1.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10057705</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBILITY ATR 72-212A 600 VERSION ATR 42-500 600 VERSION</td>
<td>Introduction of AFIRS 228.</td>
<td>STC Twenty One Limited Flight Manual Supplement No. S21.FMS-0128 Issue3 STC Twenty One Ltd Master Document List S21.MDL-1259 Issue 1 or later revisions of the above listed documents approved by EASA.</td>
<td>02/10/2018</td>
<td>Active</td>
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<td>10057706</td>
<td>SAAB AB</td>
<td>REV. 1</td>
<td>NH-500D 369HM, 369HS 360D, 369E, 369H, 369HE</td>
<td>Increase payload of Data Acquisition Module - Hardware, DAIM-H, Vairon Mount Module Installation</td>
<td>The Type and OSD Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval within...</td>
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<td>10057722</td>
<td>HELIKOPTER AIR TRANSPORT GESELLSCHAFT Am.b.H</td>
<td>REV. 2</td>
<td>EC135 T2+, EC135 T3 EC135 T2, EC135 T3 EC135 P3, EC135 T1, EC135 T2, EC135 T2, EC135 T2, EC135 P2+ EC135 P1, EC135 P3, EC135 P1, EC135 P2</td>
<td>EO 23.011: Tactical Radio Installation Description: Subjected to this modification is a standarized and modernized Tactical Radio Installation for the Christophorus Fugittungswein and EC135 feat. Therefore, the existing approvals EASA R.C. 02649, EASA R.C 02705 and EASA MCA 10047429 are combined a...</td>
<td>The Certification Basis for the original product as amended by the following: additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-27 Amdt 3: CS27.21, CS 27.25, CS 27.27, CS 27.29, CS 27.301, CS 27.303, CS27.305, CS 27.561, CS 27.601, CS2...</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 21A,109 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/10/2020</td>
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<td>Limitations/Conditions</td>
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<td>10057727</td>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>REV. 1</td>
<td>FAA STC ST0420AT</td>
<td>A330-200 SERIES</td>
<td>Interior Reconfiguration.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10057729</td>
<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINE S.A.R.L.</td>
<td></td>
<td>D2A0-A1 / D2A0-C1</td>
<td>AVIONICS MODIFICATION (installation of G3X).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>To be installed according with SEAM 4146 I Rev 7 and SEAM 4146 IP rev. 4. To be operated according with Flight manual supplement SEAM 4146 AFMS03x Rev.3. To be maintained according with SEAM 4146 ICA rev. 4.</td>
<td>Limited to VFR-Day. Further limitations are in the AFMS. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/04/2016</td>
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<td>N.S.E.</td>
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<td>SA227-TT</td>
<td>737-800</td>
<td>Modification of radio electrical install law. Dossier 45/2008</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>NSE Doc 45/2008 MDL, Rev. 3NSE Doc 45/2008 AFMS GNS S507 TAWS, Rev. 3NSE Doc 45/2008 AFMS GNS 430V AFMS Doc 45/2008 AFMS SN 4550, Rev. 3later revisions of the above listed documents approved by EASA.</td>
<td>11/04/2016</td>
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<td>10057746</td>
<td>TURK HAVA YOLU TECHNIK A.S. (TURKISH TECHNIC Inc.)</td>
<td></td>
<td>737-800</td>
<td>Cabin Retrofit by seat and partition panel installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Turkish Technic Master Document List No. CE-2500015- MD01-00 revision 0 dated 19 February 2016.</td>
<td>Restricted to MSNs 29773 to 29776, 29779, 29780, 29790.</td>
<td>11/04/2016</td>
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<td>10057746</td>
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<td>10057755</td>
<td>AVIATECH &amp; CONSULTING, SOCIEDAD LIMITADA</td>
<td>REV. 3</td>
<td>R172K</td>
<td>CESSNA R172K</td>
<td>Modification for skydiving operation.</td>
<td>The Type and OSD Certification Bases for the original product are amended by the following: additional or alternative airworthiness requirements Special Condition (s): SC-230-39-01 Parachuting Activity.</td>
<td>SAFM P01-15 TAE AFMS For Parachuting Activities in Cessna R172K, Rev. For later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2022</td>
<td>Active</td>
<td>10057755</td>
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<td>10057764</td>
<td>FLYING COLOURS CORP.</td>
<td>REV. 3</td>
<td>TCAC C-LSA16-004D</td>
<td>BD-700-1A10</td>
<td>Installation, Aircraft Interior (Special Mission/Provisions) Addition of aircraft S/N 60030 and 60034/Revision 1 - Installation, Aircraft Interior (Special Mission/Provisions)(s)</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.</td>
<td>For aircraft s/n: 9714, 9743, 9787, 9803, 60034 B/E Aerospace Limited Modification Data Summary K172000 Rev. 5, dated 16January 2023 B/E Aerospace Limited Aircraft Flight Manual Supplement K172000 Rev. 1 of 20 March 2020 B/E Aerospace Limited Mandatory Aircraft Maintenance Manual Supplement...</td>
<td>Limited to aircraft S/N: 9714, 9743, 9787, 9803, 60034, 60111 and 60118. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/06/2024</td>
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<td>Certificate Basis (CB)</td>
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<td>Limitations/Conditions</td>
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<td>Status</td>
<td>System Record</td>
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<td>P 180 AVANT II AVANTI</td>
<td>The Certification Basis for the original product as amended the following: additional or alternative airworthiness requirements: paragraph(s) at a later amendment: CS 23 Amptos. IS/1-1, 29, 301, 303, 305, 307, 337, 341, 365, 561, 571, 573, 575, 601, 603, 605, 607, 608, 611, 613, 625, 627...</td>
<td>- Airbus FMS 610099 Rev. A, dated 08.04.2016 Master Data List MSL-800304 Rev. A, dated 08.04.2016.</td>
<td>EASA STC no.</td>
<td>02/02/2021</td>
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<td>TCCA STC SH14-46</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td></td>
<td>FAA STC</td>
<td>13/04/2016</td>
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<td>10057770</td>
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<td>AS 360 B, AS 350 B1, AS 350 B2, AS 350 B3</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
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<td>FAA STC</td>
<td>13/04/2016</td>
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<td>AS 350 B, AS 350 B1, AS 350 D, AS 350 B2, AS 350 B3</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
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<td>FAA STC</td>
<td>13/04/2016</td>
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<td>AVH-1000-403, Version A</td>
<td>The Certification Basis for the original product as amended the following: additional or alternative airworthiness requirements: paragraph(s) at a later amendment: CS/23 Amptos. IS/1-1, 29, 301, 303, 305, 307, 337, 341, 365, 561, 571, 573, 575, 601, 603, 605, 607, 608, 611, 613, 625, 627...</td>
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<td>FAA STC</td>
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<td>AS 350 B</td>
<td>Aerotec Concept Modification No. 45528. The Certification Basis for the original product as amended the following: additional or alternative airworthiness requirements: paragraph(s) at a later amendment: CS/23 Amptos. IS/1-1, 29, 301, 303, 305, 307, 337, 341, 365, 561, 571, 573, 575, 601, 603, 605, 607, 608, 611, 613, 625, 627...</td>
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<td>AS 350 B</td>
<td>Additional or alternative airworthiness requirements: CS/23 Amptos. IS/1-1, 29, 301, 303, 305, 307, 337, 341, 365, 561, 571, 573, 575, 601, 603, 605, 607, 608, 611, 613, 625, 627...</td>
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<td>FAA STC</td>
<td>The Certification Basis for the original product as amended the following: additional or alternative airworthiness requirements: paragraph(s) at a later amendment: CS/23 Amptos. IS/1-1, 29, 301, 303, 305, 307, 337, 341, 365, 561, 571, 573, 575, 601, 603, 605, 607, 608, 611, 613, 625, 627...</td>
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<td>FAA STC</td>
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<td>FAA STC</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>CONSTANT AVIATION, LLC</td>
<td>FAA STC SA0356AT</td>
<td>EMB-505 Installation of Anriat ATG-4000 High Speed Internet System (AMSS) and/or Aviatrix II Tridum Satcom or Aircell Aviator 200 SSBB Satcom, AirCell ATG - 5000 High Speed Internet System (AMSS) all with Wireless Local Area Network (WLAN).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirement for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>1. FAA Approved Airplanes Flight Manual Supplement, Document number OB-1000000 06, Revision A dated April 5, 2013 or later FAA approved revision 1.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/02/2016</td>
<td>Active</td>
<td>10057792</td>
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<td>VIH AEROSPACE Inc.</td>
<td>TCCA STC SH14-43 ISSUE 2</td>
<td>412 (ASN 33001 THRU 33107) 212 Replacement of OEM Master Caution Panel in accordance with Document MDLR212-1311-MDL</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>- Certification Plan Report Number V212-1311-CP, Revision 1 dated June 17, 2014 - Compliance Record DAR333 24-CR-01 Rev. 1 dated January 8, 2016 - Compliance Record DAR 23.573, 23.303, 23.305, 23.307, 23.337, 23.341, 23.385, 23.561, 23.571, 23.573, 23...</td>
<td>The approval is applicable on Bell 212 model and on Bell 412 model to S/N 33001 thru 33107. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/05/2016</td>
<td>Active</td>
<td>10057794</td>
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<td>RUAG AG</td>
<td>REV. 2</td>
<td>P.180 AVANTI II AVANTI</td>
<td>Piaggio P180 Avanti Upgrade Received - 2x Collins VHF 20C COM transceiver and control head CTL-20C. 2x Collins VHR 32 NAV receiver and control head CTL-32. 1x Universal UNI-H FMU processor unit, display/ control unit, amplifiers and GPS antenna. 1x Collins RTA582 weather radar incl. control panel a...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 23.301, 23.307, 23.325, 23.337, 23.341, 23.385, 23.561, 23.571, 23.573, 23...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/02/2021</td>
<td>Active</td>
<td>10057812</td>
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<td>EMBRAER ENGINEERING &amp; TECHNOLOGY INC TER USA, INC.</td>
<td>FAA STC SA0303CE</td>
<td>EMB-505 Installation of an Aviator 200 SATCOM System. This STC was originally issued as a validation of Brazilian CST (STC No. 20145090-02 (3 September 2014), and transferred to a United States based Design Approval Holder on 17 July 2019.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>- Flight Manual Supplement: Doc. No. FMS-61089 Rev. B - Instructions for Continuous Airworthiness: Doc. No. ICA-8030002 Rev. A - Master Data List: Doc. No. MDL-8003039, Rev. B - Engineering Order: Doc. No. EO-8000569, Rev. B - Declaration of compliance: Doc. No. DOC-8003138, Rev. B - Master Data List: Change Approval Shown ref: CAS-A320-15-001 0 Rev.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/11/2019</td>
<td>Active</td>
<td>10057814</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>A320-232</td>
<td>NILE AIR Cabin rearrangement. The modification consists in cabin rearrangement from 180 passengers (Economics) to 164 (8 Business + 156 Economics)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>- Master Data List: Change Approval Shown ref: CAS-A320-15-001 Rev 0 - Instruction for Continued Airworthiness, ref. ICA-A320-15-015-001 Rev 0</td>
<td>According to the design related to the STC, this modification must be installed in conjunction with STG Aerospace (BAeTIgo) Installation Instructions ref: SG-320-SSL32003 photoluminescent emergency floor path marking system installation approval under existing EASA (STC no: 10049566 rev 1). Prior...</td>
<td>18/02/2016</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10057628</td>
<td>TOST GmbH</td>
<td>REV. 1</td>
<td></td>
<td>NIMBUS-4T</td>
<td>SM 03-2020 Installation of TOST hydraulic brake system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation of TOST hydraulic brake system in accordance with TOST GmbH Document SM 03-2020, initial issue, dated December 16, 2020. Instructions for Continued Airworthiness, TOST GmbH document SM003-2020, initial issue, dated December 16, 2020 or later revisions of the above listed document (s) a) - Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2020</td>
<td>Active</td>
<td>10057628</td>
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<td>10057845</td>
<td>AVONIK STAUBLING ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td></td>
<td>P210N, P210R</td>
<td>Installation of TAS Antennas.</td>
<td>The Certification Basis for the original product as amended/ threfollowing additional or alternative airworthiness requirements:CS-23, Amendment 4301, 303, 305, 307, 365, 561, 571, 575, 605, 607, 609, 611, 613, 625, 843, 867, 1301, 1306, 1308, 1309, 1327, 1351, 1431, 1529 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Project Description ASR-2016-001-PRD-04-00, Amend. 00Instructions for Installation ASR-2016-001-PRD-04-00, Amend. 00Instructions for Continued Airworthiness ASR-2016-001-ICA-11-01 (Rev.01) dated 27th March 2017for later revisions of the above listed documents approved by EASA.</td>
<td>30/09/2017</td>
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<td>10057873</td>
<td>BOGGI S.R.L.</td>
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<td>AS 365 N3</td>
<td>- (Change 365P005-00)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Flight Manual Supplement 365P005-05-00, Issue 1 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>2/04/2016</td>
<td>Active</td>
<td>10057873</td>
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<td>10057883</td>
<td>CERTIFIED TECHNOLOGY, INC</td>
<td></td>
<td>FAA STC</td>
<td>SEE FAA AML</td>
<td>Installation of Certified Technology, Inc. quick release door pins.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certified Technology Drawing list CT9801 Revision New Release, dated June 3, 1989. Certified Technology Drawing list CT9801 Revision New Release, dated June 3, 1989. 1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. 2. This STC is valid for aircraft types and models listed in the above listed documents.</td>
<td>22/04/2016</td>
<td>Active</td>
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<td>10057899</td>
<td>AEROTEC &amp; CONCEPT</td>
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<td>EC 120 B</td>
<td>MOD 14285 – Installation of Aspen 1500H (PFD and MFD)</td>
<td>The Certification Basis for the original product as amended/ threfollowing additional or alternative airworthiness requirements:The following paragraph(s) at a later amendment: CS-27, Amndt. 3: 27.24, 27.25, 27.30, 27.31, 27.305, 27.375, 27.561, 27.603, 27.607, 27.610, 27.771, 27.773, 27.13...</td>
<td>RFM SupplementV5.0 – F-500/Type Airtech helicopters EC120 Bv1.1 – Installation Aspen Evolution 1500H (PFD and MFD)Doc n° AE14269RFMS, Rev 0-0, dated 29th March 2016for later revisions of the above listed documents approved by EASA. Certification Programme Aspen Evolution 1500H (PFD and MFD)install...</td>
<td>- Use limited to Day-VFR. No connection with autopilot/Photo installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/04/2016</td>
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### List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
<th>STC Ref.</th>
<th>STC Date/No.</th>
<th>Certification Basis</th>
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<th>Date of Issue</th>
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<tr>
<td>Gulfstream Aerospace Corporation</td>
<td>10057926</td>
<td>FAA STC STD0204AT-D</td>
<td>Synthetic Vision System (SVS)</td>
<td>Approval</td>
<td>26/04/2016</td>
<td>Active</td>
<td>10057925</td>
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<td>Aerodata AG</td>
<td>10057927</td>
<td>B300</td>
<td>TCAS Software Upgrade 7.1</td>
<td>Installation, Operation and Maintenance according to the “MasterDocument List” number MDL-AC-884 Rev. 2 dated 26th April 2016 or later revisions of the above listed documents approved by EASA.</td>
<td>26/04/2016</td>
<td>Active</td>
<td>10057927</td>
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<td>Standard AER (FAA STC STD0204CH-D)</td>
<td>10057930</td>
<td>MF900, F9003EX</td>
<td>Installation of Honeywell OUI-875 Display</td>
<td>Airworthiness Limitations Supplement ref. 1019267 revision A, dated 17th March 2011.</td>
<td>27/04/2016</td>
<td>Active</td>
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<td>Air Plains Services Corporation</td>
<td>10057937</td>
<td>FAA SA0303W</td>
<td>Installation of McCauley D3A34C401/9000FA.-2, -4, -6, -8, and -10pnapeller.</td>
<td>The installation of McCauley D3A34C401/9000FA.-2, -4, -6, -8, and -10pnapeller is limited to: - not over 84 inches in diameter for float planes - not over 84 inches in diameter for land planes.</td>
<td>29/04/2016</td>
<td>Active</td>
<td>10057937</td>
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<tr>
<td>Luft Hansa Technik AG</td>
<td>10057951</td>
<td>SEE APPROVED MODEL LIST (AIRL.)</td>
<td>Installation of photoluminescent Exit Signs at Bathula or Claxton Div.</td>
<td>- Supplemental Type Definition Document - STCDD3-1219M issue 1 rev 1 ICA.</td>
<td>-</td>
<td>29/04/2016</td>
<td>Active</td>
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**Notes:****
- **STC Holder**: The company or entity that holds the Supplemental Type Certificate.
- **STC Date/No.**: The date and number of the Supplemental Type Certificate.
- **Certification Basis**: The basis for the certification, which may include reference to applicable regulations or standards.
- **Associated Technical Documentation**: Documentation related to the certification basis.
- **Limitations/Conditions**: Any limitations or conditions associated with the certification.
- **Date of Issue**: The date the Supplemental Type Certificate was issued.
- **Status**: The current status of the Supplemental Type Certificate, such as Active.
- **System Record**: A unique identifier for the Supplemental Type Certificate.

**EASA** is the European Union’s Agency for Safety in Aviation. For more information, visit easa.europa.eu.
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<th>STC Holder</th>
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<th>Foreign/NAAA STC ref.</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES N.C.</td>
<td>767-300</td>
<td>10057954</td>
<td>10057958</td>
<td>VIP CABIN MODIFICATION S-25-4035 in SABENA PA-28-181 (ARCHER III) PA-28-180 (ARCHER II) PA-28-180 (CHEROKEE) PA-28-161 (WARRIOR III) PA-28-180 (CHEROKEE) PA-28-151 (CHEROKEE) PA-28-150 (CHEROKEE) PA-28-140 (CHEROKEE CRUISER)</td>
<td>Equipment/Furnishings - Passenger</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Summary. Northwest Aerospace Technologies, Inc. document No. N543-D00. Revision 1, dated 22 April 2016 - Instructions for Continued Airworthiness, Health/Tecnic document No. HTM1033 (+v.1) Revision 1, dated 08 March 2016</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/04/2016</td>
<td>Active</td>
<td>10057954</td>
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<tr>
<td>ICELANDAIR EHF</td>
<td>767-300</td>
<td>10057954</td>
<td>10057958</td>
<td>INSTALLATION OF ZODIAC NULOOK INTERIOR ON ICELANDAIR EHF</td>
<td>10057974</td>
<td>Modification Approval Sheet MA5-S-25-4035 Issue 1 dated 29.04.2016</td>
<td>28.04.2016</td>
<td>Master Data List MDL-S-25-4035 Issue 1 for later revisions of the above listed documents approved by EASA.</td>
<td>29/04/2016</td>
<td>Active</td>
<td>10057954</td>
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<tr>
<td>SABENA TECHNICS BOD</td>
<td>A310-308</td>
<td>10057958</td>
<td>10057960</td>
<td>VIP CABIN MODIFICATION S-25-4035 in accordance with Master Data List MXD-S-25-4035 Iss. A</td>
<td>PA-28-181 (ARCHER II) PA-28-180 (ARCHER II) PA-28-180 (CHEROKEE) PA-28-161 (WARRIOR III) PA-28-180 (CHEROKEE) PA-28-151 (CHEROKEE) PA-28-150 (CHEROKEE) PA-28-140 (CHEROKEE CRUISER)</td>
<td>Replacement of Wheel and Brake System Revision 4: Addition of PA-28-140 model/Revision 5: Update with new servicing instructions within ICA document and addition of SansAIR device as option</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Modification File: DM-STC-014 rev 07 dated 11 Mar. 2022: Installation and Instructions for Continued Airworthiness: ICA- STC-014 rev 07 dated 19 Oct. 2021 to later revisions of the above listed documents approved by EASA.</td>
<td>02/06/2016</td>
<td>Active</td>
<td>10057958</td>
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<tr>
<td>BERGERING AERO</td>
<td>REV. 5</td>
<td>10057958</td>
<td>10057960</td>
<td>206L-3, 206L-4, 407 206B, 206B, 206L, 206L-1</td>
<td>Installation of the HELICOM V2/V2+ ANA Flight Data Monitoring System Revision 1 Add Bell 407 model/Revision 2: STC holder name change from I.S.E.I International Servizi ad to SAFRAN ELECTRONICS &amp; DEFENCE BUSINESS UNIT - HELICOPTERS DATAMANAGEMENT SOLUTIONS &amp; SERVICES</td>
<td>Installation of the HELICOM V2/V2+ ANA Flight Data Monitoring System Revision 1 Add Bell 407 model/Revision 2: STC holder name change from I.S.E.I International Servizi ad to SAFRAN ELECTRONICS &amp; DEFENCE BUSINESS UNIT - HELICOPTERS DATAMANAGEMENT SOLUTIONS &amp; SERVICES</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (v). CSR F-01 &quot;Umbrella Battery Installation&quot; dated 18/04/2019. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable</td>
<td>Modification File: DM- STC-014 rev 07 dated 11 Mar. 2022: Installation and Instructions for Continued Airworthiness: ICA- STC-014 rev 07 dated 19 Oct. 2021 to later revisions of the above listed documents approved by EASA.</td>
<td>18/03/2022</td>
<td>Active</td>
<td>10057960</td>
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<tr>
<td>SAFRAN ELECTRONICS &amp; DEFENCE BUSINESS UNIT - HELICOPTERS DATAMANAGEMENT SOLUTIONS &amp; SERVICES</td>
<td>REV. 2</td>
<td>10057958</td>
<td>10057960</td>
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<td>18/03/2022</td>
<td>Active</td>
<td>10057960</td>
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<tr>
<td>Q.C.M. DESIGN GmbH</td>
<td>525</td>
<td>10057958</td>
<td>10057960</td>
<td>ADS-B Out Upgrade for a Cessna 525, Removed: Garmin GTX-330D Transponder/Refitted: Garmin GTX-330D w/EIS Transponder/Compliance to CS-ACNLD A30 and AMC 20-24 was shown.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-ACNLD A30/CS-ACNLD A30B/CS-ACNLD A30C The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Airplane Flight Manual Suplement: Doc. No. C255-QCM04-001-M1-AMF, Rev 0.0 Instructions for continued Airworthiness: Doc. No. C255- QCMM34-001-M1-ICA, Rev 0.0 revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/06/2016</td>
<td>Active</td>
<td>10057958</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System</td>
<td>Description</td>
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<tr>
<td>STC Holder</td>
<td>Introduction of an Oil Dispersant Fluid</td>
<td>09/05/2016</td>
<td>Active</td>
<td>System</td>
<td>The Certification Basis for the original product as amended the following or alternative airworthiness requirements: Applicable regulations at CS 25 Amendment 4 (refers to the associated Certification Plan 4959-01-A-CPR-01, section 58, 6), for details. The requirements for environmental protection.</td>
<td>10058034</td>
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<tr>
<td>GVH AEROSPACE</td>
<td>Introduction of Cobham Aviator 300D SATCOM System with IGA-5001 Antenna</td>
<td>06/05/2016</td>
<td>Active</td>
<td>System</td>
<td>The Certification Basis for the original product as amended the following or alternative airworthiness requirements: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>10058030</td>
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<tr>
<td>CAA UK RESPONSIBILITY</td>
<td>Installation of frameless seat cushions in accordance with Helifab Inc. Master Drawing List, Document No. HF-MDL-7-1, Rev. 14, dated 1 September 2015 and FAA approved revision Rotork Gt.b and System Supplement.</td>
<td>18/05/2016</td>
<td>Active</td>
<td>System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>10058036</td>
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<tr>
<td>AIRBORNE SCIENTIFIC INC.</td>
<td>Installation of camera pod on Cessna series 172, 180 and 185.</td>
<td>11/05/2016</td>
<td>Active</td>
<td>System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>10058041</td>
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<td>2 EXCEL DESIGN LIMITED</td>
<td>Introduction of a Oil Dispersant Fluid Storage and Delivery System (TERPSUS), project: JN491-005 Rev. 1: Addition of Rhoamine B and Kilfrost DF Plus fluids to TERPSUs System, project: JN030-01 Rev. 2: Kilfrost DF Plus fluid replaced with Monomer Glycol Glycol project 2XLL05 Rev. 3: Removal of existing AFM.</td>
<td>09/07/2020</td>
<td>Active</td>
<td>System</td>
<td>The Certification Basis for the original product as amended the following or alternative airworthiness requirements: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>10058044</td>
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<td>SAFRAN CABIN INC.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>11/05/2016</td>
<td>Active</td>
<td>System</td>
<td>The Certification Basis for the original product as amended the following or alternative airworthiness requirements: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>10058047</td>
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<tr>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>LEARJET Inc.</td>
<td>FAA STC ST00136MC</td>
<td>BD-700-1A10</td>
<td>Installation of an ADS-B Out Inhibit capability on a Bombardier BD-700-1A10 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Master Document List BAS7015011, revision (G), dated April 12, 2016. FAA approved April 21, 2016. 2. Instructions for Continued Airworthiness, BAS70150601-ICA-1, revision(B), dated April 13, 2016. FAA approved April 21, 2016. 3. FAA approved Airplane Flight Manual Supplement, AFMS BAS701450601,Rev-...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/05/2016</td>
<td>Active</td>
<td>10058054</td>
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<tr>
<td>STG AEROSPACE GERMANY GmbH</td>
<td>REV. 2</td>
<td>757-200, 757-300</td>
<td>B757 – LiteMod Cabin LED Light V1.1 Installation Rev. 1: Transfer from Aviation Traders Limited (via ATL) to AviationTraders Limited Europe SAS. Rev. 2: Transfer from ATL aerospaceEurope to STG Aerospace GmbH.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>ATL Master Document List No: MSD-12992-00001 Issue 7&lt;(*)(&gt;) or later revisions of the above listed document(s) approved/ accepted for the type list under the EASA system.</td>
<td>Applicable to all Boeing 757-200/-300 aircraft fitted with DiNel, Bouchier Page ceiling and recessed lights, and Grimeca type Sidewallfluorescent lighting as defined in ATL Modification Summary SheetMOD-12992-00001 Prior to installation of this change, repair it must be determined that the interrelationships...</td>
<td>28/09/2021</td>
<td>Active</td>
<td>10058067</td>
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<tr>
<td>CRANFIELD AEROSPACE LTD</td>
<td>CAA UK RESPONSIBILITY</td>
<td>JETSTREAM 4100</td>
<td>Replacement of Smoke Detectors.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Cranfield Aerospace Ltd Type Design Definition Addendum No CAAD/ DR0323b Issue B</td>
<td>None</td>
<td>10/06/2016</td>
<td>Active</td>
<td>10058068</td>
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<tr>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 2</td>
<td>A340-642, A340-643</td>
<td>Installation of Gogo 2KU – System Activation Rev. 2: Removal of restriction in Flight Manual Supplement.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The following additional or alternative airworthiness requirements: CS 25 Amendment 16: 25.23 ALL; 25.29 (b); 25.301 (a)(1)(i)(k); 25.303 ALL; 25.305 (a,b)(i)(k); 25.307 (a)(i); 25.321 (j); 25.331 (a,b)(i); 25.333 (a).</td>
<td>- STC TWENTY ONE LIMITED Aircraft Flight Manual Supplement S21.44-68-1009, approved by EASASTC 10055030, must be installed prior to, or concurrently with, this design change. There are Airworthiness Limitations in the form of structural inspections required for this design change which are documented within S21.TEC-0497.</td>
<td>STC TWENTY ONE LIMITED Design change S21.44-68-1009, approved by EASA. STC 10055030, must be installed prior to, or concurrently with, this design change. There are Airworthiness Limitations in the form of structural inspections required for this design change which are documented within S21.TEC-0497. The...</td>
<td>22/11/2018</td>
<td>Active</td>
<td>10058069</td>
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<td></td>
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<tr>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 1</td>
<td>747-400</td>
<td>Virgin Atlantic Airways B747-400 Cabin Reconfiguration/Decrease capacity of Upper Class seats (UCS) (5) and increase both the Premium Economy (WE) and Economy (Y) cabin seating. Changed seat configuration from 44J/62W/26Y to 14J/64W/37Y. Installation of new IFE/Panasonic eXk in j-class and W-class seat...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>AFMS: S.OAEZ.001459- AFMS Rev 009ML S.OAEZ.001459-149ML Rev 009CA: S.OAEZ.001459-149ML Rev 009CA: S.OAEZ.001459-149ML Rev 009CA: S.OAEZ.001459-149ML Rev...</td>
<td>Prior to installation of this repair it must be determined that the interrelationship between this repair and any other subsequently installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/04/2021</td>
<td>Active</td>
<td>10058077</td>
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<tr>
<td>SERAM - SOCIETE D'ENTRETIEN ET REPARATION AERONAUTIQUE MACONNAISE</td>
<td></td>
<td>A36</td>
<td>FM416R Revision 2 dated 09 May 2016 - INSTALLATION GPS DS-4A304VR/ FORBRNAV, PRNAV, RNP APCH operations</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 23 initial issue added in the FACE FM416R Rev2. The requirements for environmental protection and the associated...</td>
<td>Supplément au Manuel de Vol Avion Revision II du 27 Avril 2016 later revisions of the above listed document(s) approved by EASA.</td>
<td>None</td>
<td>12/06/2016</td>
<td>Active</td>
<td>10058089</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10058091</td>
<td>PRATT &amp; WHITNEY ENGINE SERVICES, INC</td>
<td>FAA STC ST11685C</td>
<td>680A (LATITUDE)</td>
<td>Installation of a Pratt &amp; Whitney Flight Data Acquisition, Storage, and Transmission System (FAD)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. FAA Instructions for Continued Airworthiness for Cassna Citation Latitude 680A, document no. GNA-T-365-4, revision A, dated 19 February 2019. 2. FAA Instructions for Continued Airworthiness for Cassna Citation Latitude 680A, document no. GNA-T-365-4, revision A, dated 12 February 2016.</td>
<td>23/05/2016</td>
<td>Active</td>
<td>10058091</td>
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<td>10058096</td>
<td>LAZ-TEC, LLC</td>
<td>FAA STC SR00792DE</td>
<td>429</td>
<td>Approval of a seating configuration to allow flight operations with the hand pilot seat removed.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Laz-Tec Drawing List, Report No. LT-14211, Rev. 2, dated 28 November 2014. 2. FAA approved December 18, 2014. 3. Flight Manual Supplement, Laz-Tec, LLC Document No. LT-14214, No Revision, dated December 12, 2014, or later FAA approved revision. 4. Instructions for Continued Airworthiness, Laz-Tec...</td>
<td>12/05/2016</td>
<td>Active</td>
<td>10058096</td>
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<tr>
<td>10058098</td>
<td>LAZ-TEC, LLC</td>
<td>FAA STC SR00791DE</td>
<td>429</td>
<td>Installation of the skid-gear mounted step to assist in external hoist operations.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. FAA Drawing List, Report No. LT-15200, Rev. 1, dated 26 April 2016. 2. FAA approved September 1, 2016. 3. Flight Manual Supplement, Laz-Tec, LLC Document No. LT-15205, No Revision, dated August 28, 2015, or later FAA approved revision. 4. Instructions for Continued Airworthiness, Laz-Tec...</td>
<td>12/05/2016</td>
<td>Active</td>
<td>10058098</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC ref.</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10058102</td>
<td>DASSAULT AVIATION BUSINESS SERVICES/ CES SA</td>
<td>MYSTERE-FALCON 50</td>
<td></td>
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<td>TCAS II change 7.1 – TTR4100</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Aircraft Flight Manual Supplement No. TAG-M353-AMF53 Rev. IR dated 04.04.2016 or later revisions of the above listed documents approved by EASA.</td>
<td>Limited to Falcon 50 EX versions. Prior to installation of this design change it must be determined if the modification/interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>12/05/2016</td>
<td>Active</td>
<td>10058102</td>
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<tr>
<td>10058130</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC STD0402AT-D</td>
<td>REV. 1</td>
<td></td>
<td>G300, G400</td>
<td>Removal of Windshield Wiper System and Application of Windshield Surface Seal Coating. This modification consists in the installation of a windshield surface seal as the primary means of rain repellent and the deactivation/removal of the existing windshield wiper blade system. Rev. 1: Addition of the FAA...</td>
<td>- Index List, ref: CE41 4681507 Revision A dated 02 March 2015 - Aircraft Flight Manual Supplement, ref: CE41 4681508 dated 02 March 2015 - Instructions for Continued Airworthiness, ref: CE41 4684032 Revision A dated 19 February 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined if the modification/interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>30/01/2024</td>
<td>Active</td>
<td>10058130</td>
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<tr>
<td>10058132</td>
<td>AVIDYNE CORPORATION</td>
<td>FAA STC SA0034B0</td>
<td>REV. 3</td>
<td></td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of a single configuration or of a single configuration/Avidyne Corporation Integrated Flight Display, Model IFD500 and/or Model IFD400, loaded with Release 10.1 software and/or Atlas, loaded with Release 10.3 software. Revision 3 adds an additional hardware part number.</td>
<td>Installation in accordance with Avidyne Corporation Aircraft Flight Manual Supplement (AFMS) Document Number 660-0028A-000, Revision 02, FAA Approved on 18 June 201...</td>
<td>Prior to installation of this design change it must be determined if the modification/interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>07/06/2023</td>
<td>Active</td>
<td>10058132</td>
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<td>10058140</td>
<td>TRACES AIRCRAFT CERTIFICATION ENGINEERING SERVICES</td>
<td>FAA STC ST048800C</td>
<td>FAA STC ST048800C</td>
<td>737-800 SERIES</td>
<td>Installation of FWD LH Overhead Bin and Interior Furnishings.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>TRACES Master Data List, TAG-1122MDL Revision B, dated April 12, 2018.</td>
<td>Prior to installation of this design change it must be determined if the modification/interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>23/06/2016</td>
<td>Active</td>
<td>10058140</td>
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<td>10058141</td>
<td>GAMA AVIATION (ENGINEERING) LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>206L-3, 206L-4, 407 206A, 206B, 206L, 206L-1</td>
<td>Installation of Garmin G500H System.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-27 Amendment 3, 27.1, 27.27, 27.29, 27.301, 27.303, 27.305, 27.307, 27.309, 27.321, 27.337, 27.561, 27.601, 27.602, 27.603, 27.906, 27.907, 27.908, 27.910, 27.911, 27.913...</td>
<td>Garmin Aviation (Engineering) Ltd Rotorcraft Flight Manual Supplements RFMS-1440 Issue 1, Bell 206LRFMS-1440-02 Issue 1, Bell 206ARFMS-1440-03 Issue 1, Bell 206L-TRMS-1440-04 Issue 1, Bell 206L-3 and 206L-4 RFMS-1440-05 Issue 1, Bell 407</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change and any previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/10/2016</td>
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<td>10058144</td>
<td>SKYTRAC SYSTEMS Ltd.</td>
<td>REV. 1</td>
<td>TCIA SH15-12</td>
<td>AW139</td>
<td>SkyTrac ISAT Airborne Position/Data Communicator Installation. Revision 1 of this STC corrects clerical errors, updates the Associated Technical Documents section and reflects the address change of the STCholder.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Avionics Design Services Ltd. Master Drawing List MDL14069-1 Rev. Y dated 08.06.2020; Flight Manual Supplement FMS14069-1 Rev. G dated 08.06.2020; Instruc...</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10058153</td>
<td>CANARD AEROSPACE CORPORATION</td>
<td>FAA STC ST4057CH</td>
<td>ATR 72-211/-212/-212A ATR 72-101/-102/-201/-202 ATR 42-200/-300/-320/-500</td>
<td>Installation of the CMA-6800 AMLCD EFIS Displays.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. New: PAGEThe requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Canard Aerospace Corporation (CAC) Document Control Drawing D-01499 Rev. 1 dated February 11, 2016, or later FAA approved revision 2; Instructions for Continued Airworthiness ICA14069-1 Rev. N dated 08.06.2020; FAA approved revision 10 dated February 2020.</td>
<td>Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10058171</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>BD-700-1A11</td>
<td>Replacement of large monitors.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI D-01/1042177 Glass Screens of Displays (Monitors).The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List MDL-DT02-4425 Rev. B dated 17.05.2016; FAA approved revision 10 dated February 2016.</td>
<td>The installation of this STC is limited to MSN 9166. Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10058175</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Installation of G500 and G700IS for DH400 aeroplanes in accordance with Aerotec Concept Master Data List AE1608/MDL at Rev. 1 dated 12.05.2016.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aerotec Concept Master Data List AE1608/MDL at Rev. 1 dated 12.05.2016; FAA approved revision 10 dated February 2016.</td>
<td>The installation of this STC is limited to MSN 9166. Prior to installation of this design change it must be determined that the installation of this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/05/2016</td>
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### List of EASA Supplemental Type Certificates

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<td>10058198</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST0180W</td>
<td>LEARJET 45</td>
<td>Installation of an Executive Interior, Floor Plan FP49-3B in the Model45, Learjet 70 series.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>1. Learjet 45 Master Document List M1005761CT, revision K, dated December 14, 2015, FAA approved December 18, 2015. 2. MDL M1005761CT-001 configuration instructions for Continued Airworthiness. M1005761CT-ICA-001, revision N, dated December 14, 2015, FAA approved December 18, 2015. 3. MDL M1005761CT-001 configuration instructions for Continued Airworthiness. M1005761CT-ICA-001, revision N, dated December 14, 2015, FAA approved December 18, 2015. 4. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10058209</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>REV. 1</td>
<td>55</td>
<td>PNRNAV Compliance in accordance with JAA TGL No. 10</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement, AG1021-071, revision A, dated 29/February 2016. Instructions for Continued Airworthiness – ICA, AG1021-086, revision IR, dated 14 December 2015.</td>
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<td>10058213</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SRO9603RC</td>
<td>412EP</td>
<td>Installation of a cockpit voice / flight data recorder (CV/FDR) and Quick Access Recorder (QAR) for Bell Helicopter Textron 412EP Model Helicopter in accordance with Master Drawing List AA-12049.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: FAA FAR 29.301(b), 29.303, 29.305(a), 29.307(a), 29.307(b), 29.309(a), 29.601(a), 29.603(b), 29.605(a), 29.605(a), 29.611, 29.619, 29.825(a), 29.1307 and 29.30-29.613 (d)…</td>
<td>- Helicopter Flight Manual Supplement AA-12051 No Revision, dated December 03, 2013. - Instruction for Continued Airworthiness AA-12050 Revision A, dated August 27, 2013, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 200404/020 or subsequent revisions…</td>
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<td>10058216</td>
<td>ELDEC LLC</td>
<td>FAA STC ST0188W</td>
<td>DHC-8-400 SERIES, DHC-8-300 SERIES, DHC-8-100 SERIES</td>
<td>Installation of Crane Aerospace &amp; Electronics SmartBlom Wireless Tire Pressure Sensing System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Duncan Aviation Inc., Document List No. 11120713, Revision A, dated July 24, 2012.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10058239</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>A330-243</td>
<td>Air Europa – Cabin rearrangement/Cabin seating rearrangements including a configuration with “STELIA Business seats and IFE adaptation.”</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- Master Data List: Change Approval Sheet ref: CAS-A330-15-012 Rev 0. - Instruction for Continued Airworthiness, ref: ICA-A330-15-002 Rev 0 or later revisions of the above listed documents approved by EASA.</td>
<td>According to the design related to this STC, This modification must be installed in conjunction with EASA STC no. 10014567 of Bell B-412SP-H Glass Cockpit Pratt &amp; Whitney Canada PT6T-9 Twin Power Helicopter. Section Turboshift Engine associated with associated systems Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2016</td>
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<td>10058279</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>58P</td>
<td>25745F156221 / Relocation of COM antenna on fuselage top. Installation of one Avidyne TAS antenna on fuselage top. Installation of one Avion TTS antenna on fuselage bottom.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- 25745F156221-0109 / Master Document List, rev. 0, dated 21.03.2016, FAA approved 26/03/2016. - 25745F156221-0107 / Instruction for Continued Airworthiness, rev 0 dated 21.03.2016, FAA approved 26/03/2016.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>757-200</td>
<td>Introduction Crew Oxygen Charging Panel</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>STC Twenty One Limited MASTER DOCUMENT LIST No. 621. MDL-1269 Issue 1.</td>
<td>None</td>
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<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>A340-311, A340-313</td>
<td>Cabin Recconfig. Turkish Airlines A340-300 change of seat configuration 34BC/29EC to 354ECRev 1.: Transfer of STC from SR Technics Suisse to groWING Engineering Partners GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AFMS: S.OAEZ.001482-AFM Revision 03ICA: S.OAEZ.001482-ICA Revision 01MDL: S.OAEZ.001482-MDL Revision 00CCS including DoC: S.OAEZ.001482-CCS Revision 00r later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<td>KLM ENGINEERING &amp; MAINTENANCE</td>
<td>777-300ER</td>
<td>Cabin reconfiguration of KLM 777-300ER aircraft</td>
<td>The Certification Basis (CB) for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-254 (Post TC special condition – an Overhead Crew Rest/Compartments) (closeout 16 April 2003, which adopted special condition FAA 25-230-85C, CRID-2).</td>
<td>Master Data List, KLM document No. MDL-013135, Revision 03, dated 3 June 2015, Instructions for Continued Airworthiness, KLM document No. ICA-777-013135, Revision ORG, dated 27 May 2015 and all later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<td>JAMCO AMERICA, INC.</td>
<td>777-300 SERIES</td>
<td>Installation of 3 to 2 class interior reconfiguration on Boeing Model 777-300 series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Jamco America, Inc. Master Drawing List (MDL) MDL1153-1, Revision A, dated March 01, 2016.</td>
<td>1. Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product. 2. Supplemental Instructions for Continued Airworthiness.</td>
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<td>TURBINE KINETICS, INC.</td>
<td>429</td>
<td>Installation of a barrier net on the Bell 429 rotorcraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Drawing 429-001-901-DEC, Revision F, dated 8 June 2015, 2. HEICO Instructions for Continued Airworthiness (ICA), Document Number HAC14-002, Revision 0, dated 4 March 2015.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product. 2. Supplemental Instructions for Continued Airworthiness.</td>
<td>10/06/2016</td>
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<td>B/E AEROSPACE, Inc. - FSI</td>
<td>737-800 SERIES</td>
<td>Installation of interior reconfiguration on Boeing Model 737-800 series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FSI Master Drawing List (MDL) B4FSA10-D01, Revision F, dated August 27, 2015.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<td>10058328</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC 10766CH-D</td>
<td>MF50, MF900, F900EX</td>
<td>F2000, F2000EX</td>
<td>Validation of FAA STC10766CH- Dimalization of Max-Viz Enhanced Vision System (EVS) Two configurations - F2000 and F2000EX With Skycoast Displays, the Skycoast Displays are mounted in the lower aft STC - 2F000, F2000EX and MF50, MF900 and F900EX with CMS Electronic Flight Bag (EFB).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>FAA Supplement ref. 050728009 revision A, dated 24 April 2009 for the certification of F2000 and F2000EX with Skycoast Displays. FAA Supplement ref. 06033045 revision B, dated 24 April 2009 for the certification of F2000, F2000EX, MF50, MF900 and F900EX with CMS Electronics CMA-1100 EFB or later revision.</td>
<td>16/06/2016</td>
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<td>10058333</td>
<td>BROMMA AIR MAINTENANCE AB</td>
<td>650 CITATION VII 650 CITATION III</td>
<td>Cesna 650 Special Mission Air Pressure Bushhead Feedthrough</td>
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<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
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<td>MCFARLANE AVIATION, Inc.</td>
<td>FAA STC SA02113AK</td>
<td>180E, A185E, A185F</td>
<td>185, 185A, 185B, 185C, 185D,</td>
<td>Fabrication and installation of Stabilizer Jack-Screw Actuator.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>North Pacific Welding Drawing Number C-0707750, dated March 31, 2000 and Cesna Skywagon 180 and 185 Series Service Manual; or McFarlane Aviation, Inc., Drawing 1111-1, revision original, dated June 1, 2010 or later FAA approved revisions.</td>
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<td>MCFARLANE AVIATION, Inc.</td>
<td>FAA STC SA02112AK</td>
<td>180K</td>
<td>180E, 180F, 180G, 180H, 180J, 180K, 180A, 180B, 180C, 180D</td>
<td>Fabrication and installation of Stabilizer Jack-Screw Actuator.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>North Pacific Welding Drawing Number C-0707750, dated March 31, 2000 and Cesna Skywagon 180 and 185 Series Service Manual; or McFarlane Aviation, Inc., Drawing 1111-1, revision original, dated June 1, 2010 or later FAA approved revisions.</td>
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<td>10058341</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>787-8, 787-9</td>
<td>MEDICAL STRETCHER INSTALLATION ON B787</td>
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<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
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<td>MCFARLANE AVIATION, Inc.</td>
<td>FAA STC SA02115AK</td>
<td>182, 182A, 182B, 182C, 182D</td>
<td>Fabrication and installation of Stabilizer Jack-Screw Actuator.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>North Pacific Welding Drawing Number C-0707750, dated March 31, 2000 and Cesna Skywagon 180 Series 1962 and prior Service Manual; dated February 1, 1962; or McFarlane Aviation, Inc., Drawing 1111-1, revision original, dated June 1, 2010 or later FAA approved revisions.</td>
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<th>Associate Technical Documentation</th>
<th>Limitations/Conditions</th>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>ST04317AT-O</td>
<td>GV</td>
<td>Installation of Honeywell Primus Elite Advanced Features (PEAF) UpgradeDU-885 Displays.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Gulfstream Aerospace Corporation Index List CES12088003, Rev: A, dated June 1, 2016, or later FAA approved revision For -501 Configuration/FAA accepted Instructions for Continued Airworthiness CES1208AD02.Revision -, dated January 7, 2015 For -503 Configuration/FAA accepted Instructions for ContinuedAirworthiness...</td>
<td>- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. Gulfstream Aircraft Service Change (ASC) 1...</td>
<td>08/06/2016</td>
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<tr>
<td>HELI-ONE (NORWAY) AS</td>
<td>REV. 1</td>
<td>AS 332 L1, AS 332 L</td>
<td>EFI-800H EFIS system installation Rev. 1: Replacement of Standby Indicator (ref. AWSE 74760)</td>
<td>The Type and OSID Certification Bases for the original product remain applicable as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 29 Ardmt. 3, dated 11.12.2012 for the following: 29.1, 29.29, 29.29.29.30.1, 29.30.29.30.5, 29.30.7, 29.30...</td>
<td>Rev 0-Product Change Declaration HN332-00103-2201, dated 27.05.2016 Certification Programme HN332-00103-2301 Rev. A dated 26.05.2016 Declaration of Compliance HN332-00103-1001, dated 27.05.2016 Heli One Aircraft Flight Manual Supplement HN332-00103-1901 date:6.03.2016 Rev 1 - Product Change De...</td>
<td>- A copy of EFI-800H Pilot's Handbook must be immediately available to the flight crew. - The EFGS EFI-800H system is not approved for Back Course Approaches Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously...</td>
<td>29/06/2023</td>
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<tr>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>TCCA STC SA14-48</td>
<td>DHC-8-314, DHC-8-315 DHC-8-301, DHC-8-311 DHC-8-201 DHC-8-202 DHC-8-106 DHC-8-102, DHC-8-103</td>
<td>Cabin LED Lighting System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Engineering Order, Monument LED Lighting System DHC-8-100200/300; Doc Nr. CE167-031, Issue 2 - Engineering Order, Ceiling and Sidewall LED Wash Lighting Installation DHC-8-100/200/300- Doc Nr. CE167-402, Issue 1 - Maintenance Manual Supplement, Aircraft Cabin LED Lighting System DHC-8-100/200/300- 06, Doc...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. Gulfstream Aircraft Service Change (ASC) 1...</td>
<td>09/06/2016</td>
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<tr>
<td>DRF STIFTUNG LUFTrettUNG EMMENHUTZE AG</td>
<td>MBB-BK 117 C-1 MBB-BK 117 B-2</td>
<td>Installation of ADELIT</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 29 Ardmt. 3, forparagraph 29.2, 29.29, 29.29.29.30.1, 29.30.29.30.5, 29.30.7, 29.30...</td>
<td>- MBB-BK 117 B-C2-1 FMS 10-DRF-032 Rev. 00, ALS section of ICA document CA-071-25-50-01019 Rev. 00 or later revisions or the above listed documents approved by EASA – Master Document List MD-071-25-00-002 Rev. 00- Compliance Checklist CC-071-25-00-009 Rev. 00</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. Gulfstream Aircraft Service Change (ASC) 1...</td>
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<tr>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>FAA STC ST04354AT</td>
<td>767-300 SERIES</td>
<td>Interior reconfiguration installation on a Boeing 767-300 Seriesaircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. HAECO Cabin Solutions Master Data List 16T009L001, revision IR, dated June 08, 2016 2. TIMCO Aerosystems Instructions for Continued Airworthiness 16T009R009, Rev. IR, dated 26 April 2018.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/06/2016</td>
<td>Active</td>
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## List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<th>Status</th>
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<tr>
<td>AVCON INDUSTRIES, Inc.</td>
<td>F28 MARK 1000</td>
<td>ABE-NG - Scoop and Deflector Installation/Addition of one scoop and one deflector to improve nose pod cooling performance</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Program_CP-FX164-21-001E-AMPS- FX164-21-004 issue A &amp; AFM Supplement-ICA-FX164-21-005 issue A &amp; CVDeclaration of Compliance: MAS-S-21-4204 issue 5 for later revisions of the above listed documents approved by EASA.</td>
<td>- Only applicable for MSN 11290-The aircraft pre-modification configuration is the ABE-NG configuration (EASA STC No. 10056116)</td>
<td>13/06/2016</td>
<td>Active</td>
<td>10056390</td>
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<tr>
<td>BELL HELICOPTER TTXTRON, Inc.</td>
<td>412EP</td>
<td>Installation of a Honeywell Mark XXII HTAWS Kit for Bell Helicopter/Texton 412EP Model Helicopters</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: FAR 29.301(c)(2)(ii)(G), FAR 29.303(a), FAR 29.305(b), FAR 29.307(a), FAR 29.601(a).</td>
<td>- Rotordraft Flight Manual Supplement AA-1209B No. 81 Issue, dated 30th September 2012 or later.</td>
<td>Installation of Bell Basis/Proff Glass Cockpit and Pratt &amp; Whitney Canada/PT6-T Twin Power Section Turbohaft Engines and associated systems in accordance with EASA STC 10057657 is a prerequisite for the Honeywell Mark XXII HTAWS Kit.</td>
<td>13/06/2016</td>
<td>Active</td>
<td>10058397</td>
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<tr>
<td>SABA TECHNICS BOD</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Installation, Special Mission Equipment</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- TCCA Statement of Compliance, ref. NAI 5810-12-001, dated 17th June 2016- Revision A 1st March 2016-L-3 Communications MAS (CANADA) Inc. Master Document List, ref. doc.RAL-411-0004, Revision A, dated 1st March 2016-L-3 Communications MAS (CANADA) Inc. Supplements/Alpine Flight Manual ref. doc. AFM-411-0001, Revision C.</td>
<td>EASA STC 10056497 Rev. 1 is a prerequisite for this installation. Applicable only For CL-604 Aircraft Serial No. 5896 and 5898. For Operating Limitations refer to Limitations Section of the Supplemental Aircraft Flight Manual ref. doc. AFM-411-0001, Revision C.</td>
<td>14/06/2016</td>
<td>Active</td>
<td>10058421</td>
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<tr>
<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>FALCON 2000EX</td>
<td>Installation of Rockwell Collins Synthetic Vision System (SVS) and upgrade of Rockwell Collins Pro Line 21 to include TAWS Mode 5 warning during LPV Approaches</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Drawing List 514-00-001C, revision C, dated 12 December 2014.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. This STC is valid for aircraft types and m...</td>
<td>15/06/2016</td>
<td>Active</td>
<td>10058425</td>
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**Notes:**
- STC Holders and Models are listed.
- Certifications and associated technical documentation are detailed.
- Limitations and conditions are specified.
- Dates of issue and statuses are indicated.

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**EASA:** European Union Aviation Safety Agency

**Website:** easa.europa.eu
The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

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List of EASA Supplemental Type Certificates

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<tr>
<td>10058463</td>
<td>PMV ENGINEERING</td>
<td>A300 F4-605R, A300 F4-622R, A300 C4-620, A300 C4-605 R VARIANT F, A300 B4-622, A300 B4-622R, A300 B4-605R, A300 B4-620, A300 B4-601, A300 B4-603</td>
<td>Dual FMS CMA-9000 activation on A300-600</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>- MASTER DOCUMENT LIST Avi-00650-MDL-R03 dated 16. Jun 2016 AIRCRAFT FLIGHT MANUAL SUPPLEMENT Avi-00650-AMFS-R02 dated 15-Jun-2016. Qualification in accordance with RTCA/DO-229D, Change 1. Minimum Operational Performance Standards for Global Positioning/System/Satellite-Based Augmentation System/ Gluten Equipment dated February 2013.</td>
<td>17/06/2016</td>
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<td>10058465</td>
<td>AVIDYNE CORPORATION</td>
<td>FAA STC</td>
<td>SA00598B0</td>
<td>SEE FAA AML REV. 9</td>
<td>Installation of Avidyne Corporation TWX 670 Tactical Weather Detection System P/N 850-00026-000 ( ), consisting of Avidyne Processor P/N70N5-0033-000- () , Sensor Systems Inc. Antenna P/N S72-17155-1 with double pla P/N S72-17150-1, and Avidyne Mounting Tray P/N700-0034-000.</td>
<td>Based on on CFR §§ 21.115 and 21.101, and the FAA policy for non-significant changes in FAA Order 8110.46, the certification bas cystallized on the Approved Model List as follows. a) The particular aircraft type certification basis...</td>
<td>1. Avidyne Corporation Document AVTWX-003, Revision 11, dated October 16, 2008, or later FAA approved revision. 2. Instructions for Continued Airworthiness (ICA), Avidyne Corporation Document AVTWX-004, Revision 07, dated June 10, 2008, or later FAA approved revision.</td>
<td>22/06/2016</td>
<td>Active</td>
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<tr>
<td>10058467</td>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>A300 F4-622 R, A300 C4-620, A300 F4-605R, A300 C4-605 VARIANT F, A300 B4-622, A300 B4-605R, A300 B4-620, A300 B4-601, A300 B4-603</td>
<td>Installation of AlliedSignal Mark II ACARS Seat Cushion and Dress Cover Replacement</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>- Master Document List, EAD Aerospace document No. 01496-EAS-A-MDL, Revision A, dated 27 April 2016. Instructions for Continued Airworthiness, EAD Aerospace document No.01496-EAS-A-ICA-01, Revision A, dated 6 April 2016 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10058474</td>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>DHC-8-314, DHC-8-315, DHC-8-321, DHC-8-201, DHC-8-202, DHC-8-102, DHC-8-103,DHC-8-106</td>
<td>Seat Cushion &amp; Dress Cover Replacement</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>- Installation of the Seat Cushion and Dress Cover Replacement must be in accordance with Transport Canada Civil Aviation (TCCA) approved Aircraft Master Data List CE 236-900 Section 1.1 Issue 8, dated December 15, 2015,- Maintenance of the Seat Cushion and Dress Cover Replacement must be in accordance with...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/06/2016</td>
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<td>10058495</td>
<td>GIVH AEROSPACE LIMITED</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>B300</td>
<td>ACES TCAS 7.0 Upgrade to TCAS 7.1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>GIVH Aerospace Flight Manual Supplement No. GIVH/0455-01-FMS-01 Issue 1 for later revisions of the above listed documents approved by EASA, GIVH Aerospace Master Data List No. GIVH-6454-01-MA-01 Issue 1.</td>
<td>None</td>
<td>20/06/2016</td>
<td>Active</td>
<td>10058495</td>
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<td>10058501</td>
<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>FAA STC</td>
<td>SA10666SC</td>
<td>C90GTI</td>
<td>Installation of upgraded hardware to the Rockwell Collins Pro Line 21/FMS-3000 to provide Localizer Performance with Vertical Guidance (LPV)</td>
<td>The Certification Basis for this installation is p/Type CertificateData Sheet 3240 plus the following Regulations that meet amendment 23-123-671Regulations that meet amendment 23-1423-711 (a) Regulations that meet amendment 23-1223-711 (a) 23-1541Regulations that meet amendment 23-3423.1, 23-1523...</td>
<td>1. Master drawing list 335-00-001, revision A, dated 12 January 2010, or later approved FAA revision. 2. Aircraft FlightManual Supplement, document 335-00-001, revision R dated 12 January 2010, or later FAA approved FAA revision.</td>
<td>1. This installation is limited to airplane serial numbers listed in BHEA Associates document 335-00-002, revision R dated 24 November 2009 or later FAA approved FAA revision. 2. Prior to installation of this design change it must be determined that the interrelationship between this design change and/or...</td>
<td>24/06/2016</td>
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<td>EASA STC no.</td>
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<td>1005851</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
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<td>A340-313</td>
<td>WLAN Upgrade/Replacement of approved wireless access points with WLAN IEEE 802.11a/b/g standard by new units with WLAN IEEE 802.11 a/n/ac standard</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval.</td>
<td>- Supplemental Type Definition Document STD203-15/27M Issue 1 Rev 0-ICA, ref A-00-16/52-ICA Issue 2 Rev 0-Compliance Record List CR/U3-15/27M Issue 1 Rev 0-AMF A-00-16/52-AMF Issue 02 Rev 0 (Internal Checking) or later revisions of the above listed documents approved by EASA.</td>
<td>As per the Instructions of Continuous Airmenworthiness, this approval for MSN 955 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon...</td>
<td>22/06/2016</td>
<td>Active</td>
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<tr>
<td>1005829</td>
<td>CMD FLIGHT SOLUTIONS, LLC</td>
<td>REV. 2</td>
<td>FAA STC ST0142ICH</td>
<td>SEE FAA AML</td>
<td>Certification of Automatic Dependent Surveillance - Broadcast (ADS-B) Out System with Rockwell Collins TDR-94D Transponders.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment. The type certification basis for the aircraft listed in the EASAP...</td>
<td>AIRPLANE FLIGHT MANUAL SUPPLEMENT, REPORT NO DB1400004 REVISION D (or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision)) and/or the Technical Implementation Procedures of EU/USA Bilateral Agreement...</td>
<td>THE TDR-94 transponder is a non-diversity transponder. Installation of the TDR-94D transponder is required for operations in accordance with European Air Operations rules. Airplane Flight Manual Supplement, Report No. DB140004 Revision C or later revision approved by EASA must be attached to the Flight...</td>
<td>18/07/2018</td>
<td>Active</td>
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<tr>
<td>1005856</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>A340-313</td>
<td>A340-313 Ka-Band SATCOM System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CRI B-2301-001 Vibration / buffeting/ compliance criteria for large external antenna installation. The requirements for environmental protect...</td>
<td>- Supplemental Type Definition Document, Ref. STD200-15/23M Issue 01 Rev 02 ICA, Ref A-00-16/51-ICA Issue 4 Rev 06 Compliance Record List, Ref. CR/20-15/23M Issue 01 Rev 02 AMF, Ref A-00-16/51-AMF Issue 07 Rev 08 or later revisions of the above listed documents approved by EASA.</td>
<td>- As per ICA Ref. A-00-16/51-ICA Issue 4 (Section 4) and AMFs Ref A-00-16-51-AMF Issue 07 (Section 2) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce...</td>
<td>23/06/2016</td>
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<td>1005840</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 12</td>
<td></td>
<td>A321-111/131-211/231 A320-211/214/-216/-222/-251/251N A319-112/114/-132</td>
<td>KA-band SATCOM System GX Connectivity Rev. 12, Revision to include additional Weight Variants in the STC databases.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CRI B-2301-002 “Vibration / buffeting/compliance criteria for large external antenna installation”. The requirements for environmental protect...</td>
<td>- Airplane Flight Manual A-00-10/45-AMF Iss/Rev 00/00 Instructions for Cont. Airmenworthiness A-00-16/45-ICA Iss/Rev 00/00. Supplemental Type Design Document STD200-15/28M Iss/Rev 10.0 Flight Crew Data A-00-16/45-FCD Iss/Rev 01/00 or later revisions of the above listed documents approved/ acceptable...</td>
<td>This STC constitutes an approval for installation only for aircraft models that satisfy the configurations and criteria in both Engineering Instructions A01Y-01-C3-E01 Rev 11 and A01Y-01-C3-E01 Rev 12 and that are listed in EAA Approved Product List. A01Y-01-C3-DR-12 Rev 14 or later revision approved/ Prior to...</td>
<td>27/09/2023</td>
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<td>1005853</td>
<td>SWS CERTIFICATION SERVICES Ltd.</td>
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<td>A330-243</td>
<td>A330-200 AirTanker Enhanced Cabin</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- SWS Certification Programme Ref. 3150-CEP-01 - SWS Declaration of Compliance Ref. SWS/C 3150 dated 24.05.2016</td>
<td>X Installation of STC2 (EASA STC 10035845) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airmenworthiness of the product.</td>
<td>27/06/2016</td>
<td>Active</td>
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<td>STC Holder</td>
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<td>TARL AEROSPACE Ltd.</td>
<td>ISSUE 1</td>
<td>EC 130 BI / EC 130 T2 / AS 350 B2 / AS 350 B3 / AS 350 B2 / AS 350 B3</td>
<td>Installation of D350-1021 Engine Cowling/ Swivel Prop Rod for ground maintenance activities on engine bay.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Installation Instructions RA-D350-1021 Revision A, dated 29 March 2016 or later TCCA approved revision. Instruction for continued airworthiness ICA-D350-1021, Revision 01, dated 22 June 2016, or later TCCA approved revision. Master Document Lists MSL D350-1021, Revision A, dated 04 April 2018 or later.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise effect upon the airworthiness of the product.</td>
<td>26/06/2016</td>
<td>Active</td>
<td>10058599</td>
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<tr>
<td>SWS CERTIFICATION SERVICES Ltd.</td>
<td>REV. 1</td>
<td>FAA STC 29302CH</td>
<td>T206H 209H</td>
<td>Installation of an Air Conditioning System, Kelly Aerospace KATS-13-001</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>1. Installation Instructions as listed on NC-13-027, revision A, dated 24 December 2013. 2. Drawing list NC-13-023, revision B, dated 24 January 2014, or later FAA approved revision. 3. FAA approved Airplane Flight Manual Supplement, Doc. No. NC-13-025, Rev. B, approved 11 February 2014, or later FAA app...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise effect upon the airworthiness of the product.</td>
<td>30/06/2016</td>
<td>Active</td>
<td>10058570</td>
<td></td>
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<tr>
<td>FALCON CREST AVIATION SUPPLY, INC.</td>
<td>FAA STC 30322CH</td>
<td>AS 350 F2, AS 355 N</td>
<td>Installation of concorde valve regulated sealed lead acid battery P/NG355.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement for American Eurocopter Model Number AS350E, F.F.1, F &amp; N with Concorde RG355, dated 9 December 1997, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise effect upon the airworthiness of the product.</td>
<td>29/06/2016</td>
<td>Active</td>
<td>10058580</td>
<td></td>
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<tr>
<td>JET AVIATION AG</td>
<td>FAA SR01186RC</td>
<td>AS 350 F2, AS 355 N, AS 350 E, AS 355 F, AS 355 F1</td>
<td>Interior Reconfiguration - Exec. Lounges, Change of the executive lounge as per Section 2 of the Certification Plan, Doc. No. JBSL-M-00529/EMR-01, Issue 4, dated 23-06-2016. The reconfiguration will not affect the passenger capacity. The interior configuration shall continue to accommodate 24 passengers...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>AFMS, Doc. No. JBSL-M-00529/EMR-01, Issue 1, dated 23.06.2016 or later revisions of the above listed documents approved by EASA.</td>
<td>Valid for MSN 38855. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise effect upon the airworthiness of the product.</td>
<td>29/06/2016</td>
<td>Active</td>
<td>10058594</td>
<td></td>
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<tr>
<td>SABENA TECHNICS BOD</td>
<td>FAA STC 30322CH</td>
<td>A330-300</td>
<td>A330 SUNLINE cabin installation/This modification installs acabin key-out, mainly composed of new seats, information signs, carpets, raceways, curtains, lateral partitions, floor pat...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>-Modification Approval Sheet, ref: MAS S-25-4072 Issue 5, dated 29 June 2016-Master Data List, ref: MSL S-25-4072, Issue 8, dated 29 June 2016; Instruction for Continued Airworthiness, ref: ICA- F3043-25-014, Issue 3, dated 21 June 2016 or later revisions of the above listed documents approved by EASA.</td>
<td>-EASA STC 1014507, Revision 2 (Installation of floor marking) maybe concurrently embodied with this modification. The In-Flight Entertainment System server (Kontrol) installed by Thales Modification (between FR26 and FR 26.1) must be deactivated and secured. The 3G/4G antenna installed by this mo...</td>
<td>30/06/2016</td>
<td>Active</td>
<td>10058601</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10058602</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SR01844AT</td>
<td>212, 412, 412EP 205A-1</td>
<td>Retroactive Hardpoints system (Fast Rope Kit) Installation in accordance with Aeronautical Accessories Master Drawing List Report No. AA-98166 and Installation Instructions Report No. AA-98167.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aeronautical Accessories Rotorcraft Flight Manual Supplement Report No. AA-15568 NR, dated 18 May 2010/0/0 later revisions of the above listed documents approved by EASA.</td>
<td>Certification of the Retroactive Hardpoints installation does not constitute operational approval; operational approval for external load operations must be granted by local aviation authorities. Prior to installation of this design change any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2016</td>
<td>Active</td>
<td>10058602</td>
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<tr>
<td>10058615</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC STG04303SE</td>
<td>787-8</td>
<td>Installation of a business class ottoman trim and finish change for new model 787-8 series airplanes</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>B/E Aerospace, Inc. - FSI (B/E - FSI) Master Drawing List (MDL B5FS542-D01, Revision A dated 29 June 2018).</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. 2. The Supplemental Instructions for Continuation...</td>
<td>05/07/2016</td>
<td>Active</td>
<td>10058615</td>
<td></td>
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<tr>
<td>10058630</td>
<td>AMAV AEROSPACE SWITZERLAND AG</td>
<td>REV. 1</td>
<td>A320-232</td>
<td>Complete interior outfitting (34 passenger VIP interior &amp; cabin crew) as defined in the Certification Plan, Doc. No. CP-EE01-2500, Rev. C dated 30.06.2016</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): - CRJ D-2350-000 Isolated Compartments, - CRJ D-2350-000 Glass Screens of Displays / Monitors, - CRJ D-2311-000 Installation of glass panels in the cab.</td>
<td>Applicable for MSN 5182 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/07/2016</td>
<td>Active</td>
<td>10058630</td>
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<td>10058637</td>
<td>ARFINN HALLAND INC.</td>
<td>TCCA O-LSA 14-0820 ISS 1</td>
<td>182Q</td>
<td>Integration of installed STC modifications, and additional utility modifications including Br(+)-V+ C25 amp standby alternator, CAM1601 starter motor and associated electrical changes, Gemsilig Muffler, and C182S Front Seats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AFMS, Doc. No. AFMS-EE01-2500, Rev. A dated 16.06.2016, ICA, Doc. No. ICA-EE01-2500, Rev. A dated 01.06.2010/0/0 later revisions of the above listed documents approved by EASA - Master Data List, Doc. No. MDL-EE01-2500, Rev. D, dated 01.07.2016.</td>
<td>Applicable for MSN 5182 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/07/2016</td>
<td>Active</td>
<td>10058637</td>
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<tr>
<td>10058643</td>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY NTS USA, INC.</td>
<td>FAA STC SAI4349AT</td>
<td>EMB-505</td>
<td>Installation of All lavatory Coat Hook Support.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Embraer Engineering and Technology Centre Master Drawing List Report No P300-008-002, Rev. B dated 13 October 2015. Instructions for Continued airworthiness in accordance with Embraer Engineering and Technology Centre Master Drawing List Report No.P300-008-003, Rev. A...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/07/2016</td>
<td>Active</td>
<td>10058643</td>
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<tr>
<td>10058644</td>
<td>ALAMO ENGINEERING GmbH</td>
<td>REV. 4</td>
<td>EMB-135R EMB-135BJ, EMB-135ER</td>
<td>Replacement of cockpit displays DUS-870 and DUS-870 with DUS-870 on EMBRAER 135 Series Aircraft Rev. 1: Editorial ChangeRev. 2: Transfer of STC from 328 Support Services GmbH to ALAMO Engineering GmbHRev. 3: Revised MDL Rev. 4: Software upgrade for DUS-870 and addition of alternate partnumber.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List (MDL)20204AT-02 Revision C, dated 22 May 2020/0/0 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/04/2023</td>
<td>Active</td>
<td>10058644</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10058690</td>
<td>ASI INNOVATION</td>
<td>REV. 1</td>
<td>B200</td>
<td>Radome Installation on Beechcraft B200 in accordance with ASI Innovation Model B200-2500-0388.</td>
<td>STG AEROSPACE LTD AEROPLANE FLIGHT MANUAL SUPPLEMENT No. SG-1005- FMS-01Rev4 A STG AEROSPACE LTD INSTRUCTIONS FOR CONTINUED AIRWORTHINESS No.SG-1005-ICA-01 Revision B STG AEROSPACE LTD Cabin Crew Data Supplement No. SG-1005-CCD-01 Revision B STG AEROSPACE LTD MMEL Supplement No. SG-1005-MMEL-01 R...</td>
<td>Applicable to Aircraft Serials BB-1270 and BB-1399 Prior to installation of this change/repair it must be determined that the installation does not affect the airworthiness of the product.</td>
<td>07/09/2016</td>
<td>Active</td>
<td>10058690</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC (if applicable)</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10058692</td>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>REV. 1</td>
<td>ST04362NY</td>
<td>A330-343</td>
<td>Installation of an external container and internal storage provisions as specified in Master Document List 01338-EAS-A-MDL.</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s): The Certification Basis for the original product as amended by the following paragraph(s):</td>
<td>Rotorcraft Flight Manual Supplement &quot;ASIS561H HELIOVUE SYSTEMS1338-EAS-A-Rmd-01 Rev for later revisions of the above listed documents approved by EASA.&quot; Instructions for continued airworthiness supplement1338-EAS-A-ICA-01, Rev. C: Compliance Checklist 01338-EAS-A-CC-01, Rev. C.</td>
<td>12/07/2016</td>
<td>Active</td>
<td>10058692</td>
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<tr>
<td>10058703</td>
<td>BOLDER MANUFACTURING LLC</td>
<td>FAAC STC ST04190CH</td>
<td>767-300 SERIES</td>
<td>Installation of Galleys F4.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Bolder Master Data List, MDL-6105090DM1, Revision NC, dated May 27, 2016.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/07/2016</td>
<td>Active</td>
<td>10058703</td>
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<tr>
<td>10058719</td>
<td>URBE AERO GmbH</td>
<td>DA-40 D</td>
<td>Synthetic Vision and ChartView / Safe Taxi</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>DA-40 DM Supplement for Integrated Avionics System Garrett's 100SYNTHETIC Vision and ChartView DOC, No.: AFM-115-01, Initial Issue: 06-Jul-2016 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/07/2016</td>
<td>Active</td>
<td>10058719</td>
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<tr>
<td>10058722</td>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>TCCA STC SA14-18</td>
<td>DHC-8-314, DHC-8-315 DHC-8-301, DHC-8-311 DHC-8-201, DHC-8-202 DHC-8-106 DHC-8-102, DHC-8-103</td>
<td>Auxiliary Power Unit Removal</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Removal of the Auxiliary Power Unit must be in accordance with Transport Canada Civil Aviation (TCCA) approved: Avmax Aviation Services (Avmax) Engineering Order CE095-401 Issue C, dated January 15, 2015, Definition of Type Design Change: Avmax Master Data List CE095-900 Issue C dated January 15, 2015 or later</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/07/2016</td>
<td>Active</td>
<td>10058722</td>
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<tr>
<td>10058731</td>
<td>SABENA TECHNICS BGC</td>
<td>EMB-145MP, EMB-145LP EMB-145LR, EMB-145LU EMB-145, EMB-145EP EMB-135LR EMB-135SU, EMB-135ER</td>
<td>Mechanical &amp; electrical installation for tablet computer support input to EMB-145 and EMB-135</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>Master Data List, ref: 0809-01-1-MDL-R01Instructions for Continued Airworthiness, ref: 0809-01-1-ICA - R01AFM ref: 0809-01-1-AFMS Rev 02r later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/07/2016</td>
<td>Active</td>
<td>10058731</td>
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<td>10058732</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 5</td>
<td>FAA STC ST04362NY</td>
<td>A330-343</td>
<td>INSTALLATION OF GOOD 202 - SYSTEM ACTIVATION</td>
<td>Rev. 4: Transfer to new STC holder Delta Engineering with FAA as Primary Certifying Authority; Rev. 5: updates the MDL, ICA and AFMS to Delta Engineering format</td>
<td>Design change D21.44-71-1010, approved by EASA STC 10059657Rev 1, must be installed prior to, or concurrently with, this design change. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/01/2024</td>
<td>Active</td>
<td>10058732</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10058735</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST0169NY-D</td>
<td>737-400</td>
<td>Installation of Panasonic Digital Multiplexed Entertainment System (DMPES). This modification installs the Panasonic DMPES/Overhead video IFE System into the Business and Economy Class cabin. This installation includes Power Supply Units video system with overhead monitors.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List 0106-O8346-2234 Revision A dated May 16, 2013. Aircraft Flight Manual Supplement, ref: 0608-O8343-2234, dated 20 may 2013- Instruction for Continued Airworthiness, ref: 0706-O8344-2234 revision A, dated 19 February 2013 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>11/02/2016</td>
<td>Active</td>
<td>10058735</td>
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<tr>
<td>10058763</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 1</td>
<td>CL-600-2E5 (RJ SERIES 1000)</td>
<td>Mechanical &amp; Electrical installations for tablet computer support in cockpit of CRJ-700/1000 and CRJ-900/900ER. CL-600-2K1 (RJ SERIES 705)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Applicable regulations at CS 25 Amendment 4 (refers to associated Certification Plan 0101-01-A-CP-002, sections 5 and 6 for details). The requirements for environmental protection...</td>
<td>Prior to installation of this design change it must be determined that the associated technical changes and/or any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/12/2016</td>
<td>Active</td>
<td>10058763</td>
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<td>10058780</td>
<td>NSE</td>
<td>AS332L1</td>
<td>A330-343</td>
<td>Installation of DTSV2 Data Tracking System V2.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment:CS83, 301-303, 305, 307, 561, 607, 609, 611, 1301, 1303, 1351, 1357, 1360-1363, 1501, 1529, 1981 at Amend. 3 for the...</td>
<td>- Rotorcraft Flight Manual Supplement RFSMS-DTSV2-AS332 Rev.02 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the associated technical changes and/or any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/07/2016</td>
<td>Active</td>
<td>10058780</td>
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<tr>
<td>10058807</td>
<td>LUTTHANSA TECHNIK AG</td>
<td>A330-343</td>
<td>Cabin Wireless DataLink System Upgrade. This modification activates an Internal Wireless DataLink system for new/Cabin Wireless Access Points supporting IEEE 802.11ac standard.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document STDG46-1612M Issue01 Rev 00 - ICA, ref A-00-15/09-ICA, Issue 1 rev 2, Compliance Record List CRL46-1612M Issue 01 Rev 05-4FM, ref A-00-15/09-ICA Issue 1 Rev 04or later revisions of the above listed documents approved by EASA.</td>
<td>According to the design related to this STC, the following STCs and modifications, must be part of the aircraft pre-required configuration and must be installed accordingly: STC1001590, Aeronca, “Assessment of Aircraft Compatibility with TPED’s on A330 aircraft” (12 May 2016b) Minor Change M...</td>
<td>15/07/2016</td>
<td>Active</td>
<td>10058807</td>
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<td>10058815</td>
<td>AVIATION FABRICATORS, Inc.</td>
<td>FAA STC SAD2RM</td>
<td>500V 550/560 5550</td>
<td>Installation of stretch or isolates.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Branson Engineering Company, Inc. Drawing list No. 90132, Revision A, dated July 2, 1975. 2. Aviation Fabricators, Inc. SIC Drawing List AF-232, Revision A, dated April 22...</td>
<td>Prior to installation of this design change it must be determined that the associated technical changes and/or any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/07/2016</td>
<td>Active</td>
<td>10058815</td>
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<tr>
<td>10058817</td>
<td>ELECTROAIR ACQUISITION CORPORATION</td>
<td>FAA STC SAD29670H</td>
<td>SEE EASA APPROVED AML</td>
<td>Installation of the Electroair Electronic Ignition System in accordance with EIS-41000 Installation Manual.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- EIS-41000 Installation Manual, Revision 3, dated June 6, 2011- FAA Approved Airplane Flight Manual Supplement Document 1015, dated July 3, 2011. Continued Airworthiness Procedures Document 1015, EA41000-3, Revision 1, dated June 7, 2011 or later revisions of the above listed documents approved by EASA in...</td>
<td>The EIS-41000 system can only replace one magneto. Prior to installation of this design change it must be determined that the associated technical changes and/or any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/07/2016</td>
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<td>STC Holder</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>CRANFIELD AEROSPACE SOLUTIONS LTD</td>
<td>CRANFIELD AEROSPACE INSTRUCTIONS FOR CONTINUED AIRWORTHINESS SERVICE/BULLETIN NO. CAA/DD/381 ISSUE APR 2014 and detail design as listed on the AML number SA04046CH, or later FAA approved revision.</td>
<td>-</td>
<td>09/02/2016</td>
<td>Active</td>
<td>10058325</td>
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<tr>
<td>L3 COMMUNICATIONS AVIONICS SYST MS, INC</td>
<td>Installation of L3 GENESIS 750 Electronic Standby Instrument System (ESI-500 / MAG-500).</td>
<td>-</td>
<td>19/07/2016</td>
<td>Active</td>
<td>10058360</td>
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<tr>
<td>NORTHERN AVIONICS S.R.L.</td>
<td>Installation of WAVE System, installation instruction.</td>
<td>-</td>
<td>20/07/2016</td>
<td>Active</td>
<td>10058449</td>
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<tr>
<td>AVIONIK STRAUBING ENTWICKLINGS GmbH</td>
<td>Installation of Garmin GT5250.</td>
<td>-</td>
<td>30/06/2017</td>
<td>Active</td>
<td>10058092</td>
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<td>AVIONIK STRAUBING ENTWICKLINGS GmbH</td>
<td>Installation of Garmin GTX 330D.</td>
<td>-</td>
<td>30/06/2017</td>
<td>Active</td>
<td>10058092</td>
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<tr>
<td>B&amp;E AEROSPACE, Inc.</td>
<td>Replacement of certain Goodrich pneumatic de-icer boots.</td>
<td>-</td>
<td>26/07/2016</td>
<td>Active</td>
<td>10058037</td>
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**List of EASA Supplemental Type Certificates**

**10058832**

**STC Holder:** CRANFIELD AEROSPACE SOLUTIONS LTD

**Associated Technical Documentation:** CRANFIELD AEROSPACE INSTRUCTIONS FOR CONTINUED AIRWORTHINESS SERVICE/BULLETIN NO. CAA/DD/381 ISSUE APR 2014 and detail design as listed on the AML number SA04046CH, or later FAA approved revision.

**Limitations/Conditions:**

- Replied to ASN 1250. The planning table seats are prohibited for use during Taxi, Take-off/ Landing. The cockpit FTI Display Units must be prohibited for use during primary flight data and must not be used for observing aircraft limitations. The cockpit FTI Display Units can only be used during parking.

**Date of Issue:** 09/02/2016

**Status:** Active

**System Record:** 10058325

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**10058836**

**STC Holder:** L3 COMMUNICATIONS AVIONICS SYSTEMS, INC

**Associated Technical Documentation:** See FAA AIM.

**Limitations/Conditions:**

- 2. FAA Approved Airplane Flight Manual Supplement (AFMS) as listed on the AML number SA04046CH, or later FAA approved revision.

**Date of Issue:** 19/07/2016

**Status:** Active

**System Record:** 10058360

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**10058849**

**STC Holder:** NORTHERN AVIONICS S.R.L.

**Associated Technical Documentation:** Installation of WAVE System, installation instruction.

**Limitations/Conditions:**

- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.1. The STC is valid for aircraft types and model variants... (continued)

**Date of Issue:** 20/07/2016

**Status:** Active

**System Record:** 10058449

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**10058922**

**STC Holder:** AVIONIK STRAUBING ENTWICKLINGS GmbH

**Associated Technical Documentation:** Installation of Garmin GT5250.

**Limitations/Conditions:**

- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

**Date of Issue:** 30/06/2017

**Status:** Active

**System Record:** 10058092

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**10058923**

**STC Holder:** AVIONIK STRAUBING ENTWICKLINGS GmbH

**Associated Technical Documentation:** Installation of Garmin GTX 330D.

**Limitations/Conditions:**

- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

**Date of Issue:** 30/06/2017

**Status:** Active

**System Record:** 10058092

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**10058937**

**STC Holder:** B&E AEROSPACE, Inc./SMR TECHNOLOGIES, INC

**Associated Technical Documentation:** Replacement of certain Goodrich pneumatic de-icer boots.

**Limitations/Conditions:**

- 1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

**Date of Issue:** 26/07/2016

**Status:** Active

**System Record:** 10058037

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**List of EASA Supplemental Type Certificates**

**STC Holder:** CRANFIELD AEROSPACE SOLUTIONS LTD

**Associated Technical Documentation:** CRANFIELD AEROSPACE INSTRUCTIONS FOR CONTINUED AIRWORTHINESS SERVICE/BULLETIN NO. CAA/DD/381 ISSUE APR 2014 and detail design as listed on the AML number SA04046CH, or later FAA approved revision.

**Limitations/Conditions:**

- Replied to ASN 1250. The planning table seats are prohibited for use during Taxi, Take-off/ Landing. The cockpit FTI Display Units must be prohibited for use during primary flight data and must not be used for observing aircraft limitations. The cockpit FTI Display Units can only be used during parking.

**Date of Issue:** 09/02/2016

**Status:** Active

**System Record:** 10058325
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<th>EASA STC No.</th>
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<th>Rev.</th>
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<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>1005938</td>
<td>B/E AEROSPACE, Inoflight SMR TECHNOLOGIES, INC</td>
<td>FAA STC</td>
<td>ST06568AC</td>
<td>DHC-8-400 SERIES, DHC-8-300 SERIES, DHC-8-200 SERIES, DHC-8-100 SERIES</td>
<td>Replacement of certain OEM pneumatic De-Icer boots.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. SMR Technologies, Inc. Pneumatic De-Icer Layout, Bombardier DC-8-100, -200, -300 Series, Document No. SMR5152, Revision 8, dated November 23, 2015, and detail design as listed on Master Drawing ListMDL-0070, Revision C, dated March 31, 2016, or later FAA approved revision 2. Pneumatic De-Icer Lay...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/07/2016</td>
<td>Active</td>
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<td>1005942</td>
<td>MT-PROPPELLER ENTWICKLUNG GmbH</td>
<td>REV. 2</td>
<td>208, 208B</td>
<td>Installation of 4-blade constant speed propeller MTV-16-1-E-C-F-R(P)/CFR250-55a on Textron/Cessna 208, 208B equipped with Pratt &amp; Whitney PT6A-42A Engine and optional increased Gross Weight. Installation of 5-blade constant speed propeller MTV-27-1-E-C-F-R(P)/CFR245-55a and optional maximum propeller sp...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>According to MT-Propeller Master Documentation List No. E-2823 Rev. B. To be operated in accordance with MT-Propeller ARM-S Document No. E-2819/Rev. 5 when the MTV-16-1-E-C-F-R(P)/CFR250-55a on the PT6A-42A engine has been installed. To be operated in accordance with MT-Propeller ARM-S Document No. E-289.</td>
<td>According to limitation section of MT-Propeller ARM-S Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/02/2019</td>
<td>Active</td>
<td>1005942</td>
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<tr>
<td>1005946</td>
<td>B/E AEROSPACE, Inoflight SMR TECHNOLOGIES, INC</td>
<td>FAA STC</td>
<td>ST03248NY</td>
<td>SAAB SF340A, SAAB 340B</td>
<td>Replacement of certain Goodrich pneumatic De-Icer boots.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. SMR Technologies, Inc. Pneumatic De-Icer Layout, SAAB 340, Document No. SMR5141, Revision 6, dated January 1, 2013, and detail design as listed on MSL-0053, Revision J, dated February 1, 2013. The Maintenance Instructions for Continuation Approval:continuation in accordance with SAAB 340 Aircraft Maintenance...</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The STC does not require revised Electric...</td>
<td>26/07/2016</td>
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<td>1005946</td>
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<tr>
<td>1005985</td>
<td>INTERNATIONAL ENGINEERINGMAS INSTANC (IEA)</td>
<td>REV. 1</td>
<td>EC135 T3(CPDG)</td>
<td>EC135 T2+</td>
<td>Installation sieges Martin Baker en cabine / Seat installation Martin Baker in cabin.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Instructions for Continued Airworthiness, document IEA-117, Issue B, dated 04/07/2106. Supplement in &quot;GOGO&quot; au manuel de vol dû de l'hélicoptère EC 135 (sous variante, document IEA-116, Issue 4, dated 04/07/2106) or later revisions of the above listed documents approved by EASA - Service Bulletin...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/12/2016</td>
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<td>1005868</td>
<td>BABCOCK MISSION CRITICAL SERVICES DESIG N A ND COMPLETIONS LIMITED</td>
<td>REV. 2</td>
<td>CAA UK RESPONSIBILI TY</td>
<td>EC135 T1, EC135 T2, EC135 T2+</td>
<td>Installation sieges Martin Baker en cabine / Seat installation Martin Baker in cabin.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS27 Amendment 2 for 27.1(a), 27.21(a), 27.251, 27.301(a), (b), 27.303, 27.305(a), (b), (c), 27.337(a), (c), 27.337(b), (c), 27.337(d), (e), 27.337(f), 27.337(g), 27.337(h), 27.337(i), 27.337(j), (k), 27.337(l), 27.337(m), 27.337(n), 27.337(o), 27.337(p), 27.337(q), 27.337(r)...</td>
<td>- Modification Letter of Certificate of Design MDL-085-014 Issue 5 - Compliance Checklist (CS27) CCL-085-003 Issue 2 - Compliance Checklist (DO-275) CCL-085-004 Issue 5 - Rollcraft Arm Supplement FSM-085-002 Issue 5 - Instructions for Continued Airworthiness ICA-085-011 Issue 4, or later revisions of the above listed documents approved by EASA - Service Bulletin...</td>
<td>The first installation of this kit must be supervised by the STC holder. Rotorcraft modified by this design change employs Night Vision Imaging System (NVIS) lighting, commonly owned NVIS lighting, which has been certified to ensure the aircraft is compatible with Night Vision Glasses (NVG). For this reason, a...</td>
<td>04/02/2020</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>10058969</td>
<td>AVONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td></td>
<td>525, 525A, 526B</td>
<td>Installation of LPV Converter</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-23, Amendment 4 (not required by 21A.101307, 581, 575, 563, 607, 611, 771, 773, 777, 867, 1301, 1309, 1311, 1322, 1327, 13...</td>
<td>- Flight Manual Supplement, ASR-2014-03-0.FMS-05-00 Amdt. 00 or later revisions of the above listed documents approved by EASA. - Project Description, ASR-2014-032-P0D-04-00 Amdt. 00 or later approved revisions - In...</td>
<td>Prior to installation of this design change it must be determined if there is an interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/death upon the airworthiness of the product.</td>
<td>29/07/2016</td>
<td>Active</td>
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<td>10058970</td>
<td>AVONIK STRAUBING ENTWICKLUNGS GmbH</td>
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<td>501, 551</td>
<td>Replacement of existing primary Attitude Indicator Display by SandelSEAS50</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-23, Amendment 3 (not required by 21A.101307, 581, 575, 563, 607, 611, 771, 773, 777, 867, 1301, 1309, 1311, 1322, 1327, 13...</td>
<td>- Flight Manual Supplement, ASR-2015-03-0.FMS-05-00, Rev. 00 or later revisions of the above listed documents approved by EASA. - Instructions for Installation ASR-2015-030IFI-10-00 Rev. 00 or later approved revisions - In...</td>
<td>A suitable attitude source and an independent standby attitude indicator must have been already installed. This STC is compatible to EASA STC 10045766 designed by Avonik/Straubing. Prior to installation of this design change it must be determined that the interrelationship between this design change and...</td>
<td>29/07/2016</td>
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<td>10058970</td>
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<td>10058981</td>
<td>SIMPLEX MANUFACTURING G CO.</td>
<td>REV. 1</td>
<td>FAA SR02351LA</td>
<td>Model 326GII Fire Attack System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- JCA Simplex 326 GII FAS document 326-102005-003, Rev. 0 dated April 2016. - Rotorcraft Flight Manual Supplement document 326-102003-03, Rev. 0 dated 21 April 2016 or later revisions of the above listed documents approved by EASA - FAA Letter 1005-15-23, dated 13 July 2016 (Statement of Compliance)</td>
<td>Prior to installation of this design change it must be determined if there is an interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/death upon the airworthiness of the product.</td>
<td>24/10/2016</td>
<td>Active</td>
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<td>10059001</td>
<td>RUAG AG</td>
<td>REV. 2</td>
<td>PC-12, PC-12/45, PC-12/47, PC-12/45A, PC-12/47E PC-12/47E</td>
<td>PC-12 Sliding Cabin Seat Installation. This change enhances the applicability of the STC to Pilatus PC-12 equipped with Overhead / drop down oxygen masks and removes the Certificate's limitation: &quot;This change is only approved for the Pilatus PC-12 with Oxygen Mask installed in the side ledge&quot;.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Flight Manual Supplement FMS-060835, Rev. 0. Instructions for Continued Airworthiness ICA-8001540, Rev. 00 or later revisions of the above listed document(s) approved/ acceptable/derivative EASA system</td>
<td>Prior to installation of this design change it must be determined if there is an interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/death upon the airworthiness of the product.</td>
<td>01/02/2021</td>
<td>Active</td>
<td>10059001</td>
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<td>10059003</td>
<td>CERTFYNATION, LLC</td>
<td>REV. 3</td>
<td>FAA ST04004CH</td>
<td>737-900, 737-900ER 737-600, 737-700, 737-800 737-600-700-800-900-9000ER</td>
<td>Activation of RAAS Functions for EGPWS/Activation of Smart Runway and Smart Landing [SRSL] Runway Awareness Advisory System (RAAS) Functions for Honeywell Mark V and Mark V-A Enhanced Ground Proximity Warning Systems (EGPWS) Rev. 1: Extends applicability to additional models Rev. 2: Includes additional M...</td>
<td>- Master Data List, Ref. CN15-005MELD-00, Revision 14-2. Airplane Flight Manual Supplement, Ref. CN15-005MFM-02, Revision 7 or later revisions of the above listed document(s) approved/ acceptable/derivative EASA system</td>
<td>- Master Data List, Ref. CN15-005MELD-00, Revision 14-2. Airplane Flight Manual Supplement, Ref. CN15-005MFM-02, Revision 7 or later revisions of the above listed document(s) approved/ acceptable/derivative EASA in accordance with the Technical Implementation Procedures of EUL USA Bilateral Agreement.</td>
<td>11/10/2023</td>
<td>Active</td>
<td>10059003</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10059025</td>
<td>DAUT AEROSPACE Ltd.</td>
<td>205A-1/212</td>
<td>SH-16-2 ISSUE 1</td>
<td>D212-1024 Main Rotor Tie Down Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List: MDL-D212-1024 Revision A, dated May 19th 2016 or later TCCA approved revision installation instructions, IH-0212-1024 Revision A, dated October 13/30/15 or later TCCA approved revision instruction for Continued Airworthiness, IGA-D212-1024 Revision G, dated October 23rd 2015 or later.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/08/2016</td>
<td>Active</td>
<td>10059025</td>
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<td>10059026</td>
<td>SAA SOLUTIONS FOR AVIATION, S.L</td>
<td>777-200LR</td>
<td>CABIN CONNECTIVITY SYSTEM IN 777</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Flight Manual Supplement S44 77702-HDK01 Edition 8, for later revisions of the above listed documents approved by EASA Service Bulletin 77702-SB001 Ed14 (22/02/15), Master Drawing List 77702-RPT02 Ed8 (10/12/15), Maintenance Manual Supplement 77702-HDK02 Ed4 (07/04/15), Wiring Diagram, Manual Supplement 77702-HDK02 Ed4 (07/04/15).</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/08/2016</td>
<td>Active</td>
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<td>10059032</td>
<td>IAC - INDUSTRIAS AERONAUTICAS DECOIMBRA, Lda.</td>
<td>U206G</td>
<td>Modification for skydiving operation, The certification basis applies to the above aircraft only.</td>
<td>IAC-C206/SH-KY-AM6 Parachuting Modification with Cessna Skydiving Components Equipment List, Issue 01 or later revisions of the above listed documents approved by EASA IAC-C206/SH-KY Master Document List, Rev. 1.</td>
<td>This modification only can be used in association with Cessna installation of skydiving components. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/08/2016</td>
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<td>10059044</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SAIO335E</td>
<td>177/177A/177B/177RG/ F177RG</td>
<td>Installation of vortex generators on the wings and tail surfaces.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Drawing Package MA2162, Revision IR, dated November 1, 2001, and Installation Manual MA2163, Revision IR, dated December 30th, 2001, or later FAA approved revision of either of these documents.</td>
<td>Approval of this change in type design applies to the above aircraft only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/08/2016</td>
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<td>10059045</td>
<td>AVOX SYSTEMS Inc.</td>
<td>FAA STC ST0141W</td>
<td>737-600-700-800-900/-900ER</td>
<td>Installation of the AVOX LavOx Passenger Lavatory Oxygen System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CRI F-113 Oxygen Fire Hazard in Gaseous Oxygen Systems (CRI F-125 Minimum Mass Flow Supplemental Oxygen/Carbon 25 Change 15 for the affected area...</td>
<td>- Master Data List, Ref. 12-286-3500-L1-00, Rev. IR Instructions for Continued Airworthiness, Ref. S5044, Rev. for later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/26/EC (or subsequent revisions of this decision) and/or the Technical Implementing...</td>
<td>As per Instructions for Continued Airworthiness, Ref. 95544/Section 11.(Prior to installation of this design change it must be determined that the above relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.)</td>
<td>05/08/2016</td>
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<td>EASA STC no.</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10059057</td>
<td>BABCOCK MISSION CRITICAL SERVICES DAN COMPLIATIONS LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>MBB-BK117 D-3M</td>
<td>NVIS Certification in accordance with BOND/MOD/126: Reason for Revision 1: Adaptability of BOND/MOD/126 extended to MBB-BK117 D-3 and MBB-BK117D-3</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s): at a later amendment: C$ 29 Adm 4: 29.129, 771, 29.173, 29.777, 29.785, 29.803, 29.853, 29.1301, 29.1309, 29, 29.1321a, 29.1321g, 2...</td>
<td>Babcock MCS Design &amp; Completions Ltd (CAA UK STC UK-00105)</td>
<td>12/01/2022</td>
<td>Active</td>
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<td>10059063</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 1</td>
<td></td>
<td>A330-203</td>
<td>AIRCRAFT T-PED TOLERANCE. Aircraft compatibility with the use of PEDs with transmitting capability from within the aircraft cabin Rev 1: The certification dossier is updated to facilitate aircraft PED's tolerance demonstration and aircraft applicability.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>Instruction for Continued Airworthiness, ref: ICAO-AP-22-16-005-001 Rev 1 &lt;(&gt;,&gt;) or later revisions of the above listed documents approved by EASA Master Data List: Change Approval Sheet ref: CAS-AP-22-16-005 Rev 4.</td>
<td>20/04/2017</td>
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<td>10059069</td>
<td>HELI-ONE</td>
<td>REV. 1</td>
<td>AS 332 L1 AS 332 L1</td>
<td>Installation of Soft Shell Interior Kit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>Manufacturing of the Soft Shell Interior Kit is to be carried out in accordance with Manufacturing Master Drawing List DWG11019-01, Rev. 0. TC.A approved February 02, 2012. Installation of the Soft Shell Interior Kit is to be carried out in accordance with Installation Master Drawing List DWG11019-01, Rev. 0.</td>
<td>EASA STC DWG11019-01-002, 02, Installed within S21.TEC-0498. It meets the requirements for the design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>11/08/2016</td>
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<td>10059071</td>
<td>INTELSAT INFILIGHT LLC</td>
<td>REV. 3</td>
<td>FAA STC STO4252NY</td>
<td>INSTALLATION OF GOOD 2KU SYSTEM ACTIVATION</td>
<td></td>
<td></td>
<td></td>
<td>EASA Approved STC TWENTY ONE LIMITED Aircraft Flight Manual Supplement S21.74-14-1008(&gt;), &lt;(&gt;,&gt;) approved by EASA STC 10055932(&lt;&gt;,&gt;) must be installed prior to &lt;(&gt;,&gt;) or concurrently with &lt;(&gt;,&gt;) or this design change. There are Airworthiness Limitations in the form of structural inspections required for this design change which are documented within S21.74-14-1008(&gt;), &lt;(&gt;,&gt;)</td>
<td>23/07/2020</td>
<td>Active</td>
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<tr>
<td>10059084</td>
<td>NSE</td>
<td>REV. 1</td>
<td>AS 206 B</td>
<td>Installation of Usage Monitoring System by means of BRITRE SAVIER, Recorders as defined in DG-0027-02-EN, Rev 17.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>Rotorcraft Flight Manual Supplement AD026 RPM/RAQ-AD26-EN, Rev 00 or later revisions of the above listed documents approved by EASA-Brite Saver 2 Addendum 002 instructions for installation of Aircrashworthiness ICA-REC-AIRRA-206. Rev 00 (no Airworthiness Limitations associated to this STC).Brite Saver 2 Mounting Instruction...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the aircraft.</td>
<td>09/06/2016</td>
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<td>10059003</td>
<td>AEERO TECHNOLOGIES LLC</td>
<td>FAA STC ST04644CH G-1158A/G-IV/GIV-SP/GIV-X</td>
<td>Installation of thermal acoustic insulation packages.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aearo Technologies Master Data List M01-1664-05014-001, Revision B dated 28 May 2015.</td>
<td>Prior to installation of this design change it must be determined whether the design change and any other previously installed design change and/or repair will introduce degradation of the original product.</td>
<td>10/06/2016</td>
<td>Active</td>
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<td>10059100</td>
<td>ICELANDAIR EHF</td>
<td>REV. 1</td>
<td>DHC-8-402 DHC-8-401</td>
<td>AFRS 220 Automated Flight Information Reporting System Installation</td>
<td>The Type and OSL Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TEN 798-47, Aircraft Flight Manual Supplement, Issue 2 dated 28/04/2016; EASA TEN 798-04-ICE, Aircraft Maintenance Manual Supplement, Issue 2 dated 04/05/2016.</td>
<td>This STC does not include the EFIS provisions of the System. Prior to installation of this change/repair it must be determined whether the design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/06/2022</td>
<td>Active</td>
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<td>10059101</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>A330-203</td>
<td>WiFi System Installation/Installation and Activation of IFE system with wireless features for VOD on T-PED</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List: Change Approval Sheet ref: CAS- A330-16-004 Rev0-Instruction for Continued Airworthiness, ref: ICA- A330-16-004-001 Rev0-</td>
<td>According to the design related to this STC, the following STC and maintenance are pre-requisites: - STC Air France, ref: CAS-A330-16-003 Rev0-Prior to installation of this design change it must be determined whether the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/06/2016</td>
<td>Active</td>
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<td>10059102</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>MBB-BK 117 D-2</td>
<td>Title: Trakka Search Light Installation/Description: Installation and operation of a Trakka A 800 searchlight on a multifunction step on the right side of the helicopter.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement, Document No. AFRM-STC- AT-094, Rev. 1, dated 20- Jul-2016</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce degradation of the original product.</td>
<td>11/06/2016</td>
<td>Active</td>
<td>10059102</td>
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<td>10059103</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>B300, B300C B200CGT, B200GT B200, B200C</td>
<td>Digital FDR and FDU Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CRI A-01 – CS 23, Amend. 3, FAR 33, Amend. 34/21</td>
<td>No.</td>
<td>None</td>
<td>11/06/2016</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<td>NOGARO AVIATION SAS</td>
<td>10059108</td>
<td>REV 4</td>
<td></td>
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<td>Installation of a ROTAX 912S on DR460 according to NOG/STC. DOSSIER DE MODIFICATION MAJEURE au STC 10059108 Adaptation pour le DR460 on a partir de 2669 » Edition 6 dated 01.06.2023</td>
<td>Certification change/repair program must be determined in the light of the interrelationship between this change/repair and any other previously installed change and/ or repair.</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/06/2023</td>
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<td>10059108</td>
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<td>ROW 44, Inc.</td>
<td>10059114</td>
<td>FAA STC</td>
<td></td>
<td>ROW 44 Wireless Broadband System with Belt Driven Compressor</td>
<td>Installation and activation of Row 44 Wireless Broadband System according to Row 44 Master Data Index List (MDIL) RWG100-0001-00 Revision L, dated 07 July 2016.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td></td>
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<td>12/08/2016</td>
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<td>10059114</td>
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<tr>
<td>ROW 44, Inc.</td>
<td>10059115</td>
<td>FAA STC</td>
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<td>BOEING 767-300 SERIES</td>
<td>Installation of a ROTAX 912iS on DR400 according to NOG/STC. DOSSIER DE MODIFICATION MAJEURE au STC 10059108 Adaptation pour le DR460 on a partir de 2669 » Edition 6 dated 01.06.2023</td>
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<td>BOMBARDIER Inc.</td>
<td>10059141</td>
<td>CL-600-2B16 (604 VARIAN)</td>
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<td>Installation of a FOxion Air Conditioning System with Bell Driver Compressor</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
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<td>17/06/2018</td>
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<td>RSG PRODUCTS, Inc.</td>
<td>10059148</td>
<td>SA 365 N1</td>
<td></td>
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<td>Installation of a HOxion Air Conditioning System with Bell Driver Compressor</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
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<td>10059156</td>
<td>ASI INNOVATION</td>
<td>Active</td>
<td>ERJ 170-200 LR</td>
<td>Embraer 175 Cabin Modification.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>ASI Innovation Master Drawing List ref. ASI-MDL-0378-01 Revision 0/1 dated 10 August 2016. ASI Innovation Service Bulletin ref. ASI-SBB-EMB175-2500-0378 Issue 0/1 dated 9 August 2016 for later revisions of the above listed documents approved by EASA.</td>
<td>For completion of the compliance demonstration to JAR 23, 25, 791 and 25, 1447 the installer should follow the instructions given in chapter 17 on “Cabin Inspection” of the document ref. ASI-SBB-EMB175-2500-0378 during the installation of this modification before the aircraft is returned to service.</td>
<td>18/02/2016</td>
<td>Active</td>
<td>10059156</td>
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<td>10059159</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>REV. 2</td>
<td>268B</td>
<td>FIREWALL MODIFICATIONThe</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraphs of a later amendment: 23.1, 23.3, 23.21, 23.25, 23.251, 23.301, 23.303, 23.305, 23.307, 23.321, 23.337, 23.341, 23.471, 23.601, 23.603, 23.605.</td>
<td>Service Bulletin No. 20832-SBB-001, Issue 2, dated 18/07/2020. Master Drawing List No. 20832-RPT0702 Issue 2, dated 16/07/2020 for later revisions of the above listed documents approved/accepted/under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/09/2020</td>
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<td>10059180</td>
<td>TELAIR INTERNATIONAL Inc.</td>
<td>FAA STC FAA-STD0438LA</td>
<td>747-8F SERIES</td>
<td>Installation of G3M and GPS Cargo Communication System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-index List, IL195100-BSGM, Rev. N/C, dated October 15 2012-Master Data List, MLD-151510D-001, Rev. B, dated October 17, 2012 for later revisions of the above listed documents approved by EASA.</td>
<td>The installation must not be incorporated in any aircraft unless EASA STC 10039641 (FAA STC ST02367LA) has been installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/06/2016</td>
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<tr>
<td>10059187</td>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>FAA STC FAA-STD0438LA</td>
<td>747-8F SERIES</td>
<td>Installation of a Rockwell Collins TCAS-4100 Traffic Alert and Collision Avoidance System in accordance with Rockwell Collins Airworthiness Design document 2100328-STD, at issue 1, dated August 24, 2016.-</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraphs of a later amendment: 23.1, 23.3, 23.21, 23.25, 23.251, 23.301, 23.303, 23.305, 23.307, 23.321, 23.337, 23.341, 23.471, 23.601, 23.603, 23.605.</td>
<td>O1495-EAS-A-ATM-01 Rev A dated 16 Mar 2016 - Aircraft Flight Manual Supplemental 1495-EAS-A-MDL, Rev E dated 19 Aug 2016 - Master Document List for later revisions of the above listed documents approved by EASA.</td>
<td>Limitation to aircraft already fitted with EAD Aerospace TC23-71-225. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/06/2016</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>10059221</td>
<td>SPAES GmbH &amp; Co. KG</td>
<td>SA 365 N1/SA 365 N1 SA 365 N2/SA 365 N3</td>
<td>Installation of adapter plate for EMS seat; For EMS operation an EMS seat has to be installed and a two seat benches has to be removed. Therefore an adapter plate as interface between the EMS seat and the ground structure of the passenger compartment has to be installed.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certificated noise and/or emissions level of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>Flight Manual Supplement “FMS-SPAES-0044”, Revision 01 or later. Revisions of the above listed documents approved by EASA Master Document List “MDL-365-044-03-01”, Revision 01, dated 17.08.2016.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change(s) or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/02/2016</td>
<td>Active</td>
<td>10059221</td>
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<tr>
<td>10059224</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERINGCO ROPERATION AG (AIEC)</td>
<td></td>
<td>Triple FMS Software Upgrade (UASC) Initial release of 10059224/Upgrade of the existing triple Flight Management System (FMS) UNG-1 Freysystem software to SCN 1001 X in accordance with Universal Avionics/System Corporation (UASC) Service Bulletin 300X.X.(34-3633, dated 07 April 2016 Rev. 1 to 10059224)...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certificated noise and/or emissions level of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>- Airplane Flight Manual Supplement. Doc. No. AC5338-071, Rev. Cor later revisions of the above listed documents approved by EASA - Master Document List. Doc. No. AC5338-049, Rev D. Engineering Order. Doc. No. AC5338-062, Rev. B. Instructions for Continued Airworthiness. Doc. No. AC5338-089, Rev. B.</td>
<td>Limited to HB-IKS, SN 5042. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change(s) or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/03/2018</td>
<td>Active</td>
<td>10059224</td>
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<tr>
<td>10059232</td>
<td>GVH AEROSPACE LIMITED</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>Installation of Thrommen HSL-1600 Searchlight and HCL-1600 Hand Control in accordance with GVH Aerospace Master Data List (MDL) - Airbus EC135 Searchlight GVH-5898-02-MDL-01.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certificated noise and/or emissions level of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>Flight Manual Supplement (FMS) – EC-135 Searchlight Installation GVH-5898-02-FMS-01 Issue 2or later revisions of the above listed documents approved by EASA - Master Document List (MDL) - Airbus EC135 Searchlight Installation GVH-5898-02-MDL-01 Issue 3Instructions for Continued Airworthiness (ICA) – EC135 S...</td>
<td>The Thrommen HSL-1600 searchlight is to be installed on an existing sidestep mounting bracket approved in accordance with Avian Camm Systems/Meeker EASA STC 10017058 (EASA.IM.R.S.01490). The minor design change designated GVH-5898-01-MDL-01 will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/08/2016</td>
<td>Active</td>
<td>10059232</td>
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<tr>
<td>10059233</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>MU-28-60</td>
<td>Installation of Garmin G6000 system with GAD 43e as an option.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval.</td>
<td>AFM supplement AISR-2016-063-FMS-05-00, amendment level 0, dated 16/06/2016 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
<td>29/08/2016</td>
<td>Active</td>
<td>10059233</td>
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<tr>
<td>10059237</td>
<td>AVIONITEC Ltd.</td>
<td>PA-60-700P</td>
<td>Avionics Upgrade Removed - Garmin GNS530(GPS/NAV/COM) with GA 56 Antenna- Garmin GNS430(GPS/NAV/COM) with GA 56 Antenna- Garmin MX 20 Multifunction Display- Eventide Argus 7000 Navigation Display- Bendix/King KI 255 HS+ Bendix/King 97 HAF with KAA448 Antenna- Bendix/King KT 76 C Transponder- RC...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment: Compliance to CG-ACNS D.ELS.010, 015, 020, 025, 030, 045, 050, 065,080. The requirements for environmental prot...</td>
<td>AFMS for dual Garmin GNS 530W/430W. Doc. No. 5846-011, Rev. 30.06.2016 AFMS for Sundair SN 3500 EHSI; Doc. No. 5846-012, Rev. 30.06.2016 AFMS for Honeywell KMH 880 TASI/TAWS; Doc. No. 5846-013, Rev. 30.06.2016 or later revisions of the above listed documents approved by EASA Master Document List. Doc.1...</td>
<td>None.</td>
<td>29/08/2016</td>
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<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>DV AVIATION, Inc; JET AVIATION AG</td>
<td>Installation of a Wing Mounted Forward Looking Infrared Camera System; Monitor Stand, and Microwave Downlink Antenna System. Rev: 3; Transfer from &quot;Fifty-Ten-Yankee, LLC&quot; to &quot;DV Aviation, Inc.&quot;</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/quartered noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>Master Certification System Safety Report No. 918-100, Revision 1, dated January 22, 2016. Pilot's Operating Handbook Supplement Safety Publication No. SA260A, Revision 2, dated September 29, 2014.</td>
<td>Approval of the change in type design applies to the above listed supplement only. Prior to installation of this design change it must be determined that the interface between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the system.</td>
<td>14/06/2023</td>
<td>Active</td>
<td>10529082</td>
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<tr>
<td>JET AVIATION AG</td>
<td>SATCOM SBB Antenna (remove 500 FC or one year limitation).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/quartered noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>INSTRUCTIONS FOR CONTINUED A-ICA-R04- AMF-350-1C Issue</td>
<td>None.</td>
<td>21/09/2017</td>
<td>Active</td>
<td>10552946</td>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>Installation of a Concorde RG-306E or RG-386E44 Lead Acid Battery. Revision 1 adds battery installation for 13TP5 models equipped with Helionix.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Except where amended by additional or later amendments, if indicated on FAA STC SR00105RBCD, the requirements for environmental protection and the associated/quartered noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>Airbus Helicopters, Inc. drawing list 135AMX-8202, Rev. F, dated 21 June 2016. For RG-386E battery installations on aircraft originally certified with the TC 25-27 Ah NICad battery and not equipped with Helionix: Rotorcraft Flight Manual Supplement (RFMS) 1SSMS-002 Rev. 3, dated 24 June 2016.</td>
<td>The RG-396E is a 28 Ampere-Hour lead acid battery and can be used as an replacement for the TC 25-27 Ah NICad battery. The RG-386E44 is a 42 Ampere-Hour lead acid battery and can be used as a replacement for the TC 25-27Ah NICad battery or the TC 41Ah NICad battery. Airbus Helicopters EC135 with Helionix.</td>
<td>10/01/2022</td>
<td>Active</td>
<td>10552927</td>
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<tr>
<td>AEROMECH Inc.</td>
<td>Group aircraft approval for Reduced Vertical Separation Minimum (RVSM).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/quartered noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Master Data list, document AM0100- DL02-00, Revision 0, dated 14 January 2016 or later FAA approved revision. Class 1 Airframes- ICA, document AM-44A, KAR-0713, Revision F, dated 1 December 2015, for FAA-accepted revision, AM-4AFSMS-KAR-0718, Revision A, dated 17 February 2016.</td>
<td>Prior to installation of this design change it must be determined that the interface between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/06/2016</td>
<td>Active</td>
<td>10552929</td>
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<tr>
<td>SABENA TECHNICS BOC</td>
<td>Mechanical &amp; electrical installation for tablet computer support incorportation of ATR 22-211/2012.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements is determined by the following paragraph(s) at a later amendment: Applicable regulations CS-25 Amendment 4 (refer to the associated Certification Plan), AM0101-A-CP- RG2, sections 5 and 6.</td>
<td>- Master Data list, ref 0661-01-A-MLD-RM- Instructes for Continued Airworthiness (Supplement), ref 0661-01- A-ICAR-44- AFM (Supplement) ref 0661-01- A-AFMS-R44 (later revision of the above listed documents approved/ accepted by the EASA system.</td>
<td>- Only for Electronic Flight Bag use. No other use permitted. In case of Electro Magnetic Interference when personal tablet computers connected to electrical installation, the power supply cord must be disconnected from the USB outlet. Its the operator's responsibility to determine that the output...</td>
<td>23/08/2018</td>
<td>Active</td>
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<td>EASA STC no.</td>
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<td>Limitations/Conditions</td>
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<td>10059204</td>
<td>PMV ENGINEERING</td>
<td>REV 2</td>
<td>A330-203</td>
<td>Cabin Reconfiguration and In Flight Entertainment System installation on A330. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. - MASTER DOCUMENT LIST SCA-0050-MDL-R03/1-10, dated 1 Sep 2016. INSTRUCTIONS FOR CONTINUED AIRWORTHINESS CAB-01811-ICA-R06/dtd. 14 Jan 2016. INSTRUCTIONS FOR CONTINUED AIRWORTHINESS SCA-0050-ICA-R01, dated 29 Aug 2016. EWS INSTRUCTIONS FOR CONTINUED AIRWORTHINESS SCA-0050-0000-R03/dtd. 29 Aug 2016.</td>
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<td>04/11/2016</td>
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<td>10059265</td>
<td>EICHELSDORFER GmbH &amp; FLUGZEUG BAU</td>
<td></td>
<td>A320 CL ASW 20, ASW 20 BI ASW 20 BL / ASW 20 C</td>
<td>Installation of the Draline PSR T01 (EASA E.108) turbine engine to the airplane ASW 20 in accordance with EichelsofFuguegebau document ASW 20 STC PSR DL, at revision 0, dated August 31, 2016.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the equivalent Safety Finding(s): 22.975, 22.977. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Start of engine based on gas. Aircraft Flight Manual Supplement, EichelsofFuguegebau document ASW20STCPSRFLG, at revision 0, dated February 18, 2016. Start of engine based on current: Aircraft Flight Manual Supplement, EichelsofFuguegebau document ASW20STCPSRFLG, at revision 0, dated duly marked</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/09/2016</td>
<td>Active</td>
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<tr>
<td>10059287</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SED0710AT</td>
<td>PT6A-27/28/34/34B</td>
<td>Installation of PC safety Valve Kit, part number 610-850-01/002.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Report No. AA-99088 Installation Instructions (P.3 Safety Valve Kit of Pratt and Whitney PT6A-270/280 series engines, Rev “A” dated/duly marked 19, 2000, or later FAA approved revisions.</td>
<td></td>
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<td>05/09/2016</td>
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<tr>
<td>10059273</td>
<td>KEIN BOREK AIR Ltd</td>
<td>TCCA SA09-59</td>
<td>DHC-6 SERIES 320 DHC-6 SERIES 310 DHC-6 SERIES 300 DHC-6 SERIES 210 DHC-6 SERIES 200 DHC-6 SERIES 110 DHC-6 SERIES 100 DHC-6 SERIES 1</td>
<td>Modification “Roof Access Step Installation” in accordance with AeroDesign Ltd. DCL857, IR dated 13 July 2009.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. DCL857, IR, dated 13 July 2009 29002986-28001253, dated 29 Aug 2016 20002986-28001253, dated 29 Aug 2016</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10059296</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td></td>
<td>A330-343</td>
<td>Interior reconfiguration of Lufthansa aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Certification Summary, Northwest Aerospace Technologies, Inc. document No. N5641-D00, Revision A, dated 31 August 2016.</td>
<td></td>
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<td>02/09/2016</td>
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<td>10059277</td>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td>A321-213/-231/-232 A321-111/-112/-113/-211/-212 A320-216/-211/-212/-233 A320-111/-112/-113/-211/-212/-213 A319/-211/-112/-113/-311/-114/-115</td>
<td>ACM5 WEFA system installation on Airbus Single Axle Aircraft The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Applicable regulations at CS 25 Amendment 4 (refer to associated Certification Plan 0549-01-A-CP-R02, section 5, for details). The requirements for environmental protection are... - Master Data List, ref : 0549-01-A-MDL-R03, Instructions for Continued Airworthiness, ref0549-01-A-ICA-R02/revisions of the above listed documents approved by EASA.</td>
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<td>STC No.</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>Model(s)</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
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<tr>
<td>10059287</td>
<td>KNOLL AVIATION ENGINEERING LIMITED</td>
<td>CA A UK RESPONSIBIL ITY</td>
<td>A330-200</td>
<td>EQUIPMENT AND FURNISHINGS – A330 TOURIST CLASS SEAT INSTALLATION</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>KNolls Aviation Engineering Limited Master Document List No. 1695871 - M DL-001-R Revision 04 dated 02 September 2016.</td>
<td>Restricted to MSNs 0587 and 0634. This STC is a prerequisite for EASA STC No. 10059284.</td>
<td>09/02/2016</td>
<td>Active</td>
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<tr>
<td>10059306</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>BAE.125 SERIES 100GA</td>
<td></td>
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<td>P-RNAV Airworthiness Approval. P-RNAV airworthiness approval based on the existing aircraft airworthiness in compliance to JAA TGL 10 rev. 1 was shown RNAV 1 compliance was shown against ICAO PBN Manual DOC 9613, TableS-3.2.</td>
<td></td>
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<td>06/09/2016</td>
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<td>10059337</td>
<td>GVH AEROSPACE SERVICES Inc.</td>
<td>CA A UK RESPONSIBIL ITY</td>
<td>A340-312</td>
<td>Installation of a 100 VIP Passenger LOPA.</td>
<td></td>
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<td>08/09/2016</td>
<td>Active</td>
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<tr>
<td>10059338</td>
<td>SOCIETE DUCh/ DUC HELICES PROPPELLERS</td>
<td>REV. 2</td>
<td>DR-400/180 S DR-400/180 R DR-400/180 T DR-300/180 R</td>
<td>Revision of STC DH400 / FLR2 – Installation Hélice 5-pales FLAIR-2H, FLR2 S-DU, CUpdate covering increase in weight, removal of the spacer, modifications of fixing configurations and addition of instruction manual in English.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Program, Doc. No. DHm STC_000 OP, Revision H, Dated 16 May 2018; - Master Document List, Doc. No. DHm STC_000 MDL, Revision C, Dated 16 May 2018 or later revisions of the above listed documents approved by EASA.</td>
<td>Limitation to aircraft fitted with LYCOMING engine O-360-A3A, A1A only. Prior to installation of this change/repair it must be determined that any replacement or repair with other previously installed design/repair and any other previously installed design change and/ or repair will not introduce noise/degradation effect upon the airworthiness of the product.</td>
<td>07/06/2018</td>
<td>Suspende d</td>
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<tr>
<td>10059339</td>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>TCCA STC</td>
<td>SA16-16</td>
<td>DHC-8-315 DHC-8-311 DHC-8-314 DHC-8-320 DHC-8-301 DHC-8-106 DHC-8-201 DCH-8-102 DHC-8-103</td>
<td>Rockwell Collins TTR-4100 TCAS II V7.1 U.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>CE329-801 Issue A, dated 11 JAN 16, Flight Manual Supplement. RockwellCollins TTR-4100 TCAS II V7.1 Upgrade.</td>
<td>Prior to installation of this design change it must be determined that any replacement or repair with other previously installed design change and/ or repair will not introduce noise/degradation effect upon the airworthiness of the product.</td>
<td>08/09/2016</td>
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List of EASA Supplemental Type Certificates

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<th>EASA STC no</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<tr>
<td>10059355</td>
<td>HAICO CABIN SOLUTIONS, LLC</td>
<td>FAA STC0198AT</td>
<td>A330-300</td>
<td>Interior Recomfiguration of A330-300.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: Refer to TIMCO Letter noTIM-16-0020, dated on 02nd March, 2016.</td>
<td>Master Data Ref 141144001, revision 0Creation plan 141144000 rev CICA ref 141144000 rev FAA Statement Of Compliance - T...</td>
<td>Refer to the associated ICA document related to the Associated Technical Documentation for this approval/Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce...</td>
<td>09/02/2016</td>
<td>Active</td>
<td>10059355</td>
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<td>10059358</td>
<td>ADS-B TECHNOLOGIES, L.L.C.</td>
<td>FAA STC SR02471AK</td>
<td>SEE FAA AML REV. 1</td>
<td>Installation of the L-3 Communications NGT-900/9000R ADS-B System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval NEW-PAGThe requirements for environmental protection and the associated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>1) Lynx NGT -9000 and CP2500 Rotorcraft Installation Manual, L-3-89-IM Revision 0, dated December 3, 2015, or later FAA approved revision.2) Rotorcraft Flight Manual Supplement or Supplemental Flight Manuals L3-89-RFM56, Revision 0, dated February 16, 2016, is required to be carried on the rotorcraft...</td>
<td>23/09/2016</td>
<td>Active</td>
<td>10059358</td>
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<tr>
<td>10059350</td>
<td>FOKKER TECHNOIEK B.V.</td>
<td>REV. 2</td>
<td>737-800</td>
<td>Boeing 737 VIP CompletionThe aircraft conversion consists of replacing and modifying the standardBoeing interior partly by installing a new VIP interior in the forward section, installing new components in the aft section and the installation and/or modification of several cabin systems.Revisions...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>AFMS: B738-RTA-AMM-001, revision See AMMS: B737-RTA-AMM-5-001, revision 03.ICA: SE-744, issue 07 MDL: B737-RTA- MDL-0001, issue 01 CRR: B737-RTA-CRR-0001, issue 02 MJM: B738-RTFAMM-5-001, revision 06 or later revisions of the above-revised document(s) approved/accepted by the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a noise/weight effect upon the airworthiness of the product.</td>
<td>05/12/2019</td>
<td>Active</td>
<td>10059350</td>
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<tr>
<td>10059387</td>
<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>REV. 3</td>
<td>HAWKER 800XP, HAWKER 850XP, HAWKER 750, HAWKER 900XP</td>
<td>Installation of a Rockwell Collins Pro Line 21 update to include optional Terrain Awareness and Warning System (TAWS) mode 5 warning/Localizer Performance with Vertical Guidance (LPV) approaches, optional Automatic Dependent Surveillance Broadcast (ADS-B) Out, optional Synthetic Vision System (SVS)...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:The following paragraph(s) at a later amendment FAR 25.319(a)(b) - Amend 25-141 FAR 25.73(a)(b) - Amend 25-109 FAR 25.85(a)(b) - Amend 25-100 FAR 25.1317(a)(b) - Amend 25-14...</td>
<td>- Installed according to the Master Drawing List 549-00-0001, Revision C, dated 21 April 2016, - Operated according to the Airplane Flight Manual Supplement549-00-0005 Initial Revision, dated 21 April 2016, - Maintained according to the Instructions for Continued Airworthiness549-00-0006, Initial Revision.</td>
<td>- Must have factory installed Proteus 21-The Database Unit (DBU) must be factory installed DBU-5010E or updated to a DBU-5010E per EASA STC Approval 10016379 (FAA STC ST/2013/01-03). If the optional LPV TAWS Mode 5 is installed, LPV must be installed form the factory, factory service bulletin or be ins...</td>
<td>18/01/2023</td>
<td>Active</td>
<td>10059387</td>
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<tr>
<td>10059374</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>FAA STC ST04511SE</td>
<td>737-800 SERIES</td>
<td>Installation of interior components.</td>
<td>The Certification Basis (CB) for the original product as amended by the following additional or alternative airworthiness requirements, the following paragraph at aileron amendment FAR 25.85(a)(b) amended 25-116. The requirements for environmental protection and the associated noise and/or emissions levels of the on...</td>
<td>Approval of this change in type design applies to the above-mentioned aircraft only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a noise/weight effect upon the...</td>
<td>19/01/2017</td>
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<td>EASA STC no.</td>
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<td>STC ref.</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>10059375</td>
<td>LIFEPORT, LLC</td>
<td>FAA STC SA0235LA</td>
<td>B200GT</td>
<td>Installation of a Patient Loading and Utility System (PLUS) seat/seat/serviPlex</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Approved Master Drawing List for the PLUS and ServiPlex installation in the Hawker Beechcraft B200GT, MLD-846-00, Revision J, dated 27 March 2009. FAA Approved Airplane Flight Manual Supplement No. EDR-846-11, dated May 13, 2009, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that there is no detrimental interaction between this design change and any other previously installed design change and/or repair work. Airworthiness and environmental protection and the associated certified noise and/or emissions level of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>15/02/2016</td>
<td>Active</td>
<td>10050372</td>
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<td>10059378</td>
<td>J &amp; C ENGINEERING SERVICES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A320-231/232/233/271N A320-211/212/214/215-216 FREE STANDING BULKHEAD</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>J &amp; C ENGINEERING SERVICES Ltd Modification No. J/C MOD/M4618 Issue 1.</td>
<td>The bulkhead installation must be in accordance with J &amp; C drawings RAC252JC181 ISSA. Subsequent installations of the divider must be the subject of separate design approval.</td>
<td>12/09/2016</td>
<td>Active</td>
<td>10050378</td>
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<td>10059379</td>
<td>AERO TECHNICS DESIGN LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A340-642/-643 A340-311/-312/-313 Introduction of alternative part number seat foams on Zodiac premium economy seats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aero Technics Design Limited Certification Plan No. ATDCP0638 Issue 3.</td>
<td>None.</td>
<td>12/09/2016</td>
<td>Active</td>
<td>10050379</td>
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<td>10059389</td>
<td>AERO TECHNICS DESIGN LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A350-341/350-342/350-343 A350-321/350-322/350-323 A350-301/350-302/350-303 Introduction of alternative part number seat foams on Zodiac Sicma economy seats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aero Technics Design Limited Certification Plan No. ATDCP0630 Issue 4.</td>
<td>None.</td>
<td>12/09/2016</td>
<td>Active</td>
<td>10050389</td>
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<td>10059391</td>
<td>AERO TECHNICS DESIGN LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A350-300 Introduction of alternative part number seat foams on Zodiac premium economy seats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aero Technics Design Limited Certification Plan No. ATDCP0632 Issue 4.</td>
<td>None.</td>
<td>15/09/2016</td>
<td>Active</td>
<td>10050391</td>
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<td>10059392</td>
<td>DELTA G DESIGN Inc.</td>
<td>FAA ST11229C</td>
<td>737-900ER Structural modification necessary for the installation of three cameras, one antenna, and one drainage hole.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: C5 25 Amendment 1 forth the affected area as defined in DGD-EASA-TC-02, Rev. JR. The requirements for environmental protection...</td>
<td>- Master Data List, Ref. DGD-SPOC-MD-01, Revision B - Instructions for Continuous Airworthiness, Ref. DGD-SPOC-IC-01, Revision A (Airworthiness Limitations). Prior to installation of this design change it must be determined that there is no detrimental interaction between this design change and any other previously installed design change and/or rework work. Airworthiness and environmental protection and the associated certified noise and/or emissions level of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Limitations as per: - Instructions for Continuous Airworthiness, Ref. DGD-SPOC-IC-01, Revision A (Airworthiness Limitations). Prior to installation of this design change it must be determined that there is no detrimental interaction between this design change and any other previously installed design change and/or rework work. Airworthiness and environmental protection and the associated certified noise and/or emissions level of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>13/09/2016</td>
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<td>STC Holder</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>OMOBADO SYSTEMS INTERNATIONAL</td>
<td>AS 350 B, AS 350 B1, AS 350 B2, AS 350 B3, AS 350 BA, AS 350 D</td>
<td>Installation of the 200-310-00 Load Weigh Kit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Onboard Systems Drawing List No. 155-116-00, Revision 0, dated December 13, 2006, or later FAA-approved revision Onboard Systems Owner's Manual No. 120-128-00, Revision 0, dated June 29, 2006, or later FAA-approved revision. Section A6 of Onboard Systems Instruction for Continuous Airworthiness...</td>
<td>Approval of this change in type design applies only to those Airbus Helicopters AS350 model rotorcraft listed above that are equipped with the Airbus Helicopter 1400 kg Cargo Swing/Lift system as installed per Eurocopter Service Bulletin 25.00.62 Prior to installation of this design change it must be deter...</td>
<td>21/02/2016</td>
<td>Active</td>
<td>10555002</td>
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<td>GFD GmbH</td>
<td>35A/ 36A</td>
<td>Decoy Self Protection System (SPS) Rev. 1: Transfer to GFD GmbH.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI F-01, Decoy Self Protection System (SPS). The requirements for environmental protection and the associated certified noise level and/or emissions level...</td>
<td>Aerospace Flight Manual Supplement, doc. 5467-01, Rev. A, dated 13 June 2016. Instructions for Continuous Airworthiness, doc. 5467-085, Rev. A, dated 13 June 2016 for later revisions of the above listed document(s) approved/ accepted by the EASA system.</td>
<td>This STC is only valid for a deactivated system, it does not grant airworthiness approval for the intended function of the Decoy system. The aircraft operator must obtain airworthiness and operational approval for the appropriate authority responsible for the type of operation involved in the system...</td>
<td>23/09/2022</td>
<td>Active</td>
<td>10555007</td>
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<td>APICAL INDUSTRIES, Inc. (d/b/a DART AEROSPAC Е)</td>
<td>MBB-BK117 C-2, MBB-BK117 D-2</td>
<td>Installation of Apical Industries, Inc. Cabin Cutters Kit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Dart Aerospace Master Document List MD145-2, Revision C, dated 12 January 2016 or later FAA-approved revision. Instructions for Continuous Airworthiness, ICA 145-21, Revision D, dated 6 June 2015, or other FAA-approved ICA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a net adverse effect upon the airworthiness of the product.</td>
<td>16/09/2016</td>
<td>Active</td>
<td>10555047</td>
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<tr>
<td>AIRCRAFT TECHNICAL DEVELOPMENT &amp; ENGINEERING COMPANY</td>
<td>737-800 SERIES</td>
<td>Installation of Cabin Interior Window Shades.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Aercon Engineering Company Master Drawing List No. AC25-WSB737-1000, Revision “K&quot;, dated 10 August 2016. 2. Aercon Engineering Company Instructions for Continuous Airworthiness (CA) Document Maintenance Manual Supplement CM-3797WS-1063 Revision &quot;NC&quot;, dated 23 June 2016.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a net adverse effect upon the airworthiness of the product.</td>
<td>16/09/2016</td>
<td>Active</td>
<td>10555046</td>
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<tr>
<td>AIM AVIATION (HENSHALLS) LTD</td>
<td>737-800</td>
<td>Installation of Galleys 1 (737GUN01GC00), 2 (737GUNF02C00), 4B (737GUNH4BG0C00) and 7 (737GUNYF5C00).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Change record CR 1175 rev. F Final Weight Report STR/FR/91175 rev Air later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a net adverse effect upon the airworthiness of the product.</td>
<td>16/06/2016</td>
<td>Suspende d</td>
<td>10555045</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>R66</td>
<td>Installation of S-TEC HeSiAS Stability Augmentation System/Autopilot (SAS/AI) in accordance with S-TEC Master Data List (MDL) ST-663-MDL-0001-0.</td>
<td>The Certification Basis (CB) for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): FAA STC No. 27-165-SC Installation of HeSiAS Autopilot and Stabilization/Augmentation System (APAS). The requirements for environmental protection and...</td>
<td>Rotorcraft Flight Manual Supplement ST-663-RFM5-0001 Revision Air later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/EF (or subsequent revisions thereof) decision and/or the Technical Implementation Procedures of EUTSA Bilateral Agreement...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a net adverse effect upon the airworthiness of the product.</td>
<td>19/09/2016</td>
<td>Active</td>
<td>10555045</td>
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List of EASA Supplemental Type Certificates

10059458
STC Holder: GAL AVIATION Inc.
Model: B737-800
Certificate basis: Installation of blade tip ref. AL II

10059468
STC Holder: SOCATA
Model: SA 3180
Certificate basis: Installation of blade tip ref. AL II

10059470
STC Holder: AVIATION TRADERS LTD
Model: 737-800
Certificate basis: Installation of blade tip ref. AL II

10059495
STC Holder: ROSEMOUNT AEROSPACE, Inc.
Model: A320-232/-233/-251N/-271N
Certificate basis: Installation of blade tip ref. AL II

10059502
STC Holder: LUFTHANSA TECHNIK AG
Model: DLH A330 2
Certificate basis: Installation of blade tip ref. AL II
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<th>EASA STC no.</th>
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<th>Certification basis</th>
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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tr>
<td>10059504</td>
<td>LITEAIR AVIATION PRODUCTS Inc.</td>
<td>REV. 1</td>
<td>FAA ST02438SE</td>
<td>A321-211/-232, A321-213, A321-232, A318-111, A318-211, A318-212, A318-213, A319-112/-113, A320-232, A320-233, A321-111, A320-212, A320-214, A320-231, A320-211/-212/-214, A319-132, A319-133, A320-211, A319-131, A319-113, A319-114, A319-115, A319-131, A319-111, A319-112, A319-113, A319-115</td>
<td>Lightweight window plug installation.Rev. 1: applicability list corrected</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/modified noise and /or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>LAP/01/2011 Rev. A: Master drawing list-LAP/01/2011 Rev. A Installation drawing including maintenance instructions and, airworthiness limitations information.</td>
<td>Prior to installation of this design change it must be determined that the installation is free of interference between this design change and any other previously installed design change and/or repair which will introduce noise effect upon the airworthiness of the product.</td>
<td>03/07/2024</td>
<td>Active</td>
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<td>10059514</td>
<td>HELI-AUSTRIA GmbH</td>
<td>REV. 4</td>
<td>D-025-001, Rev. 0, dated 12.05.2016. Installation drawing</td>
<td>A3 332 L</td>
<td>Installation of a KANNAD 406 AF-H Emergency Locator Transmitter.</td>
<td>The Certification Basis for the original product as amended to follow the requirements for Special Condition (s): CRF F-01(c), (d)</td>
<td>Master Document List MDL-13078-0001, Issue 4/12: One later revisions of the above listed document(s) approved and accepted under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the installation is free of interference between this design change and any other previously installed design change and/or repair which will introduce noise effect upon the airworthiness of the product.</td>
<td>28/09/2021</td>
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<td>10059514</td>
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<td>10059517</td>
<td>AIRBUS S.A.S.</td>
<td>A320-214</td>
<td>Airbus Corporate Jet Centre Modification CJ2070 – CABIN COMMUNICATION SYSTEM UPGRADE</td>
<td>A320-214</td>
<td>The Certification Basis for the original product as amended to follow the requirements for Special Condition (s): CRF H-02</td>
<td>Flight Manual Supplement (FMS), Doc. No. HAUD-025-405, Rev. 0, dated 07.05.2016.</td>
<td>Prior to installation of this design change it must be determined that the installation is free of interference between this design change and any other previously installed design change and/or repair which will introduce noise effect upon the airworthiness of the product.</td>
<td>22/09/2016</td>
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<td>10059517</td>
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<td>10059536</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>A320-214</td>
<td>SEE APPROVED MODEL LIST</td>
<td></td>
<td>Replacement/upgrade of the existing single or dual transponder transceiver ADS-B functionality according to Minimum Operational Performance Standards.</td>
<td>- Modification Approval sheet CJ2070 issue 01, dated 23rd September 2016: AFM Supplement (ALM) CJ2070 dated SEP 01/16 for later revisions of the above listed documents approved by EASA.</td>
<td>This approval is limited to MSN 3199</td>
<td>23/09/2016</td>
<td>Active</td>
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List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<th>Rev. level</th>
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<th>Date of Issue</th>
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<td>10059542</td>
<td>SABENA TECHNICS BGC</td>
<td>A320-213,</td>
<td>8019-01-A-MDL-R00</td>
<td>A320-213</td>
<td>Ku-Band Antenna Provision Removal on A320</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>EASA STC 10059542</td>
<td>27/09/2016</td>
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<td>10059543</td>
<td>JET AVIATION AG</td>
<td>REV. 1</td>
<td>02/09/2017</td>
<td>A319-131,</td>
<td>HD Tail, Quad and Zoom Camera Installation Revision 1 - Transfer of the EASA STC 10059543.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>EASA STC 10059543</td>
<td>20/09/2017</td>
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<td>10059546</td>
<td>AVIONIK STRAUSS ENTWICKLINGS GmbH</td>
<td>MBB-BK117 C-1, MBB-BK117 B-1, MBB-BK117 B-2</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of single/twin Garmin GTN650/GTN750.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 29, AMD 3: CS 29.207(a)(h)(J) CS 29.207(b)(h)(k) CS 29.207(c)(h)(k) CS 29.207(d)(h)(k) CS 29.207(e)(h)(k) CS 29.207(f)(h)(k) CS 29.207(g)(h)(k) CS 29.207(h)(h)(k) CS 29.207(i)(h)(k) CS 29.207(j)(h)(k) CS 29.207(k)(h)(k)</td>
<td>- Approved Flight Manual Supplement for COMNAV/GS/PS/GTIN650/GTN750 for rotorcraft equipped with Bendix/King EF50/50, AR-14-081 FMS-05.1-00. Revision 0 dated 27 September 2016. - Approved Flight Manual Supplement for COMNAV/GS/PS/GarminGTN650/GTN750 for rotorcraft equipped with Thales SM4D.</td>
<td>With this installation, the approval status of the helicopter of type B-737 persists LPV, VNAV/LNAV isn't approved. Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will...</td>
<td>27/09/2016</td>
<td>Active</td>
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<td>10059551</td>
<td>JUASPP ENGINEERING SERVICES, L.L.C.</td>
<td>FAA 5399633RC</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Ballast Pump Restraint System (BPRS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Approval of Flight Manual Supplement for COMNAV/GS/PS/GarminGTN650/GTN750 for rotorcraft equipped with Bendix/King EF50/50, AR-14-081 FMS-05.1-00. Revision 0 dated 27 September 2016. - Approved Flight Manual Supplement for COMNAV/GS/PS/GarminGTN650/GTN750 for rotorcraft equipped with Thales SM4D.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>28/09/2016</td>
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<td>10059561</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>B260GT</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Antenna installation provision for mission system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, doc. no. AE16149/CDP Rev. 0, dated 26 September 2016.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>28/09/2016</td>
<td>Active</td>
<td>10059561</td>
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<tr>
<td>10059562</td>
<td>CESINA DUESSELDORF CITATIONSERVIC E CENTER</td>
<td>525, 525A</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation provisions of Apple iPad Air series (single/Dual).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>According to Master Document List 2015-8-16, Rev. R (To be operated in accordance with AFM-S Document No. 2015-8-16, Rev A) or later revisions of the above listed documents approved by EASA.</td>
<td>Operational approval to use as EFB class 2 separate device needed. Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>28/09/2016</td>
<td>Active</td>
<td>10059562</td>
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Source: easa.europa.eu
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<th>STC Holder</th>
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<tr>
<td>CESSNA DUESSLEDORF CITATIONSERVIC E CENTER GmbH</td>
<td>525, 525A</td>
<td>Installation of Power FLARM Core.</td>
<td>According to Master Document List 2015-10-16, Rev IPTs to be operated in accordance with AFM-S Document No. 2015-10-8, Revision later revisions of the above listed documents approved by EASA To be installed i.e. Installation Procedure 2015-10-1, IF, or later Revisions To be maintained i.e. MM-S 20...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/02/2016</td>
<td>Active</td>
<td>10059653</td>
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<tr>
<td>LUTTHANSA TECHNICK AG</td>
<td>737-700</td>
<td>Ka-Band SatCom Antenna</td>
<td>- Supplemental Type Definition Doc STD003-15/20M issue 02- AFM Supplement A-00-15/26-AFM Issue 03- ICA Doc A-00-15/26-ICA Issue 04- MEP Supplement A-10-16/26-MPD Issue 02or later revisions of the above listed documents approved by EASA</td>
<td>Due to a relocation of the GPS antennae on top of the fuselage, the aircraft has lost its approval for all GNSS Landing System (GLS) functions. Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise previously installed design change...</td>
<td>02/02/2017</td>
<td>Active</td>
<td>10059657</td>
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<tr>
<td>FLYING COLOURS CORP.</td>
<td>CL-600-2B16</td>
<td>Cabin Reconfiguration.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Restricted to ASN 550L.</td>
<td>30/09/2016</td>
<td>Active</td>
<td>10059600</td>
<td></td>
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<tr>
<td>A.A.D AEROSPACE AIRWORTHINESS</td>
<td>A319-132, A319-133 A319-114, A319-115, A319-131 A319-111, A319-112, A319-113</td>
<td>Installation of KA Band Satcom radome with provisions for LRU antenna.</td>
<td>- Master Drawing List, EAD Aerospace document No. 01486-EAS-A, MDL, Revision D, dated 30 September 2016. Instructions for Continued Airworthiness Supplement, EAD Aerosopace/document No. 01486-EAS-A-ICA-01, Revision C, dated 30 September 2016: Aircraft Flight Manual Supplement, EAD Aerospace document No.</td>
<td>This modification applies only to aircraft being fitted with SR/NST. This modification applies only to aircraft fitted with SAT TV tailwindantenna, in accordance with Airbus Service Bulletin SB 56JC,014: Prior to installation of this design change it must be determined that the interrelationship between...</td>
<td>30/09/2016</td>
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<tr>
<td>JET AVIATION AG</td>
<td>A319-115</td>
<td>VVP cabin interior - EASA STC Approval. The VVP interior has been completed allowing for these convertible configurations. Baseline Configuration and Option A and Option B. Revision 3: Transfer EASA STC 10058612 Rev. 2.</td>
<td>See appendix JBSC-M-00054MC-01, Issue 04, The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/09/2017</td>
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<td>SAFRAN POWER USA, L.L.C.</td>
<td>AB139, AW139</td>
<td>Installation of starter-generator. Goodrich PN 23078-029 andDaishin Aircraft-Delegation (QAD) kit PN 23078-500.</td>
<td>Goodrich drawing 23078-029, revision D, dated July 29, 2013, Goodrich drawing 23078-500, revision A, dated December 12, 2013, later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/10/2016</td>
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<td>STC Holders</td>
<td>Rev. level</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EAGLE CPTORS USA, INC.</td>
<td>FAA SR004255E</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of a satellite phone.</td>
<td>See EASA Approved Model List. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
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<td>04/10/2016</td>
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<tr>
<td>I.S.E.I INTERNATIONAL SERVICE ELECTRONIQUE INFORMATIQUE</td>
<td>REV. 1</td>
<td>TCCA SA00-5S</td>
<td>DHC-8-315 DHC-8-311, DHC-8-314 DHC-8-302, DHC-8-310 DHC-8-106, DHC-8-201 DHC-8-102, DHC-8-103</td>
<td>Installation of SAFETYPLANE V5 system/installation of the Flight Data Monitoring and Transmitting System<em>SAFETYPLANE V5</em>system.</td>
<td>The Certification Basis for the original product as amended is the following: additional or alternative airworthiness requirements: Special Condition (s): CRI F-01 (HFHF), CRI F-03 (EL), CRI F-16 (LPO). The requirements for environmental protection and the associated/certified noise and/or emissions level.</td>
<td>AEROPLANE FLIGHT MANUAL SUPPLEMENT: N11-DA42, SPVS-E-A-01 Rev.04. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/10/2016</td>
<td>Active</td>
</tr>
<tr>
<td>AVMAX AVIATION SERVICES inc.</td>
<td>REV. 1</td>
<td>TCCA SA00-5S</td>
<td>DHC-8-315 DHC-8-311, DHC-8-314 DHC-8-302, DHC-8-310 DHC-8-106, DHC-8-201 DHC-8-102, DHC-8-103</td>
<td>GPS/SATCOM Antenna Installation</td>
<td>The Certification Basis for the original product as amended is the following: additional or alternative airworthiness requirements: Special Condition (s): CRI F-01 (HFHF), CRI F-03 (EL), CRI F-16 (LPO). The requirements for environmental protection and the associated/certified noise and/or emissions level.</td>
<td>AEROPLANE FLIGHT MANUAL SUPPLEMENT: N11-DA42, SPVS-E-A-01 Rev.04. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/10/2016</td>
<td>Active</td>
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<tr>
<td>LUFTNASHA TECHNik AG</td>
<td>737-700</td>
<td>Cabin Interior Completion.</td>
<td>The Certification Basis for the original product as amended is the following: additional or alternative airworthiness requirements: Special Condition (s): CRI F-01 (HFHF), CRI F-03 (EL), CRI F-16 (LPO). The requirements for environmental protection and the associated/certified noise and/or emissions level.</td>
<td>Supplemental Type Definition Document STD000-152/1M Issue 1.</td>
<td>Supplemental Type Definition Document STD000-152/1M Issue 1.</td>
<td>As per AFM supplement A-00-1648-AMF.Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/10/2016</td>
<td>Active</td>
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<tr>
<td>VECTOR AEROSPACE INTERNATIONAL LTD</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>AS 332 L</td>
<td>Mission System Fit (VAIL Project 160).</td>
<td>The Certification Basis for the original product as amended is the following: additional or alternative airworthiness requirements: Special Condition (s): CRI F-01 (HFHF), CRI F-03 (EL), CRI F-16 (LPO). The requirements for environmental protection and the associated/certified noise and/or emissions level.</td>
<td>Flight Manual Supplement 160-20-02 'Vector Aerospace Modifications', Release A.</td>
<td>This STC is applicable to helicopter AS 532L, S/N 2046 only. This STC provides for access to stretcher attachment points in the cabin. These attachment points are for provision only and approval for installation of stretchers on these points is not included in this STC. Installation of stretchers or any...</td>
<td>05/10/2016</td>
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<tr>
<td>STC No.</td>
<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
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<tr>
<td>10059650</td>
<td>PEREGRINE LLC</td>
<td>FAA STC</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Replacement of Honeywell (formerly Sperry) EI-800 cathode ray tube (CRT) display, part number 700315-601 or 700311-902, with Esterline CM/3 Electronic MCA-6800 multi-purpose liquid crystal display (LCD).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None.</td>
<td>06/10/2016</td>
<td>Active</td>
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<tr>
<td>10059690</td>
<td>MID-CONTINENT INSTRUMENT Co., Inc.</td>
<td>FAA STC</td>
<td>SA0075SE</td>
<td>Installation of Mid-Continent lithium battery in accordance with Master Data List AM0229-DL02-00</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>MDL AM0229-DL02-00, Rev. A dated 21-Feb-2013; FAA STC AM0229-FMA01-00, Rev. A dated 21-Feb-2013; FAA STC IC01-60, Rev. A dated 15-Jan-2013; FAA STC later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EASA Biennial Agreement E-E15-0001 Rev. C, CMA-6800 Multi-Purpose Liquid Crystal Display (LCD).</td>
<td>06/10/2016</td>
<td>Active</td>
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<tr>
<td>10059680</td>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY EMB-500</td>
<td>see Approved Model List</td>
<td>Installation of L3 Communication System Devices Inc. (L3) Com/3 Electronic MCA-6800 Multi-purpose Liquid Crystal Display (LCD).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the origin...</td>
<td>- Master Document List Number TR0127, Revision L, dated 17 December 2018; Airplane Flight Manual Supplement (AFMS), Document Number ST0321CE-A; Revision 9, dated 1 June 2009; Instructions for Continued Airworthiness (ICA), Document Number CA-002, Revision B, dated 21 September 2017; FAA STC later revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>26/11/2019</td>
<td>Active</td>
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<tr>
<td>10059670</td>
<td>SABENA TECHNICS BGC</td>
<td>ATR 72, 212A, ATR 72-212A</td>
<td>GPS/SBAS antennas structural installation on ATR42 and ATR72/212 Rev 1; include the configuration of the aircraft fitted with second GNSS antenna.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, ref: 20052-01-A-MDL-R01, Instructions for Continued Airworthiness (Supplement), ref: 20052-01-A-ICA-R01; or 2014/04/CF (or subsequent revisions of this decision)</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>27/03/2017</td>
<td>Active</td>
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<tr>
<td>10059652</td>
<td>J &amp; C ENGINEERING SERVICES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>see Approved Model List</td>
<td>Major change to STC 10059652/ICAM Installation on Starboard (on ATR42) to allow installation of an L-3 Communication System Devices Inc. (L3) Com/3 Electronic MCA-6800 Multi-purpose Liquid Crystal Display (LCD).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. J/N/A=C/E, C Engineering Services Ltd. Flight Manual Supplement CFA-FM0048 Issue 1, dated 2005-19 covers both original Starboard wing strut installation of XCAM pod and includes updates to address alternate port strut installation of XCAM Pod plus wider Cassini 172 type approval of this STC...</td>
<td>None.</td>
<td>14/08/2019</td>
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**List of EASA Supplemental Type Certificates**

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<th>STC Holder</th>
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<td>AVMAX AVIATION SERVICES Inc.</td>
<td>10059748</td>
<td>Active</td>
<td>10059748</td>
<td>DHC-8-311/210/315/316/319/320/321/322/323/324/325/330/331/335DHC-8-102/105/106/108</td>
<td>FLYHT AFIRS 228 Installation. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement CE322-801, Issue B dtd. 02 August 2010 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada/Bilateral Agreement/Maintenance Manual Supplement CE322-701, Issue A dtd. 12 November 2015 Certificate.</td>
<td>This system is not eligible for an operational approval for Long Range Communications Systems (LRS).EASA STC 10059823, Rev. 1 or later must be accomplished together with this STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and...</td>
<td>07/10/2016</td>
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<td>10059694</td>
<td>Rev. 1</td>
<td>10059693</td>
<td>AVMAX AVIATION</td>
<td>SuperSeal UltraLite (SSUL) MARKING SYSTEM EMBRAER SERIES AIRCRAFT Revision 1, Transfer of STC holder from STG Aerospace Limited to STGAerospace GmbH</td>
<td>EASA STC no.</td>
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<td>12/02/2016</td>
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<tr>
<td>STG AEROSPACE GERMANY GmbH</td>
<td>10059706</td>
<td>REV. 1</td>
<td>10059694</td>
<td>ERJ 190-100, ERJ 190-200</td>
<td>Installation of Universal UNS-1D FMS with P-RNAV capability.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Cessna Dusseldorf Citation Service Centre GmbH Master Document List ref 2016-13-16, Rev IR, dated 21 April 2016, and Engineering Order ref 2016-13 Rev IR, dated 21 April 2016. Operate in accordance with Cessna Dusseldorf Citation Service Centre GmbH Aircraft Flight Manual...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness of the product.</td>
<td>10/10/2016</td>
<td>Active</td>
<td>10059719</td>
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<tr>
<td>CESSNA DUESSELDORF CITATION SERVICE CENTER GmbH</td>
<td>10059728</td>
<td>REV. 1</td>
<td>10059693</td>
<td>DHC-6 SERIES 300 DHC-6 SERIES 200 DHC-6 SERIES 100</td>
<td>Installation of Universal UNS-1D FMS with P-RNAV capability.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation of EASA STC 10057449 (STC SA12-13) is required Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness...</td>
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<td>11/10/2016</td>
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<tr>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>10059748</td>
<td>REV. 2</td>
<td>10059694</td>
<td>DHC-6 SERIES 300 DHC-6 SERIES 200 DHC-6 SERIES 100</td>
<td>Fuel Quantity Indicating System Replacement</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Engineering Data Summary E110300, Rev. An&lt;=&gt;, Modification Instruction E110310, Rev B or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada/Bilateral Agreement/Maintenance Manual Supplement, E110321, Rev A...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/weight effect upon the airworthiness...</td>
<td>14/06/2017</td>
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<td>10059751</td>
<td>A31 INNOVATION</td>
<td></td>
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<td>DA 42 M-NG</td>
<td>Installation of Removable Protection Kit</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Flight Manual Supplement ASI-AMF-0270 Issue 02 dated September 2016 or later revisions of the above listed documents approved by EASA.</td>
<td>Limitation to Diamond DA42-M-NG serial number 007.</td>
<td>12/10/2016</td>
<td>Active</td>
<td>10059751</td>
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<tr>
<td>10059756</td>
<td>AEROTEC GROUP</td>
<td>S.A.S. transferred to 309218 with 707360</td>
<td>MBB-BK 117 C-2</td>
<td>Installation of Removable Protection Kit P/N BL1-145-01 including floorplates and seals shields.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>ATG-MV-BJ1-145-01 Rev 01 &quot;BL1-145-01 Flight Manual Supplement&quot;, dated 11/10/2016 ALS as defined in Chapter 09 of “User’s Manual” ATG-DTU-BL1-145-01 Rev 02, dated 12/10/2016 or later revisions of the above listed documents approved by EASA. ATG-DCAD-BL1-145-01 Rev 03 &quot;Declaration of Conformity and Com...</td>
<td>None.</td>
<td>12/10/2016</td>
<td>Active</td>
<td>10059756</td>
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<tr>
<td>10059760</td>
<td>DART AEROSPACE Ltd.</td>
<td>FAA STC SH15-44 ISSUE 1</td>
<td>206L/206L-1/206L-3/206L-4/206L-4/407</td>
<td>Installation of air filter D407-1007-011 PA110 PUNet System at engine intake.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement FMS-D407-1007 Rev. B, dated 05/10/2016 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement. Installation Instructions IN-D407-1007 Revision C, dated September 29, 2015 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this product.</td>
<td>13/10/2016</td>
<td>Active</td>
<td>10059760</td>
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<tr>
<td>10059765</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>757-200, 757-300</td>
<td>STC Twenty One Change 321.34.01-1600 - Introduction of ADS-B Out</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement S21.34.01-1160 - Introduction of ADS-B Out</td>
<td>None.</td>
<td>13/10/2016</td>
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<td>10059765</td>
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<td>10059766</td>
<td>365 AEROSPACE DDS LIMITED</td>
<td>FAA STC FD047-1007 ISSUE 1</td>
<td>365 Aerospace Ltd MDL-0074</td>
<td>Change for the introduction of Satellite Communications System (SATS) and Cabin Local Area Network (Wi-Fi) Rev. 1: Transfer of STC from 365 Aerospace Ltd. to 365 Aerospace DDS Ltd.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>365 Aerospace Ltd Flight Manual Supplement No. 1 FMS-0074-01 REV B, dated 12 October 2018 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other pre-existing installed design changes and repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/08/2019</td>
<td>Active</td>
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<td>10059775</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST000894Y</td>
<td>737-700/700-800/900ER</td>
<td>Installation of a Panasonic Global Communication System (GCS)/This modification installs the Panasonic GCS with the functional feature/Connect (internet access) and Cell Modern (ground maintenance connection).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Master Data List 0106-07787-2204 Revision C, dated 21 October 2015: Aircraft Flight Manual Supplement, ref. 0606-07792-2204 Revision C, dated 24 September 2015 - Instruction for Continued Airworthiness, ref. 0706-07793-2204 Revision C, dated 30 November 2015 or later revisions of the above listed documentation.</td>
<td>None.</td>
<td>14/10/2016</td>
<td>Active</td>
<td>10059775</td>
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<td>EASA STC No.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10059777</td>
<td>ESID</td>
<td>REV. 2</td>
<td>EC635 P2+, EC635 T1, EC635 T2+, EC135 T1, EC135 T2, EC135 P1, EC135 P2, EC135 P2+</td>
<td>Installation of HD ELECTRO OPTICAL SYSTEM (EOS) on EC135 (H135) Initial Rev.: Introduction of EOS of type FLIR STAR SAFIRE 380-HD and FLIR STAR SAFIRE 380-HDC on Standard Landing GearRev. 1: Extension to use High Landing Gear for EOS of type FLIR STAR SAFIRE 380-HDRev. 2: Introduction of EOS of type He...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unarchived and remain applicable to this certificate/approval.</td>
<td>- Rotordraft Flight Manual Supplement 0026-DU-MZ- FMS-00, Issue 03- Instruction for Continued Airworthiness 0026-DU-MZ-ICA-00, Issue 03- For Initial Rev.: List of Compliance Documents, 0026-DU-MZ- LCCD-00, Issue 01- For Rev. 2.:</td>
<td>04/06/2019</td>
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<td>10059778</td>
<td>PRECISE FLIGHT, Inc.</td>
<td>FAA STC 007335E</td>
<td>DHC-8-400 SERIES DHC-8-300 SERIES DHC-8-200 SERIES DHC-8-100 SERIES</td>
<td>Installation of Precise Flight Pulsetlite System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unarchived and remain applicable to this certificate/approval.</td>
<td>Precise Flight, Inc. Engineering drawings 29/F0000, Revision A, dated May 25, 2001, or later FAA approved revision- FAA approved Aircraft Flight Manual Supplement for d...</td>
<td>Approval of this change in type design applies to the above model aircraft only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon th...</td>
<td>14/10/2016</td>
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<td>10059779</td>
<td>TRUATLANTIC MFG., LLC</td>
<td>FAA STC SA02422AT-D</td>
<td>T210K,T210L,T210M,T21 ON T210K, T210L, T210M, T21ON</td>
<td>Installation of a Stainless Steel Exhaust System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unarchived and remain applicable to this certificate/approval.</td>
<td>- Atlantic Aero, Inc. Master Drawing List 20060301, Rev. E, dated November 20, 2006 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/00 (or subsequent revisions to this decision) and/or the Technical Implementation Procedures of EUUSA Bilat.</td>
<td>1. The following models are eligible for installation of this STC: a. 210K, L, M, N, Serial Numbers 2105920 through 21064857 with the Teledyne Continental Motors (TCM) IO-550-L engine, or those incorporating STC’s which install the IO-550-P engine b. T210K, L, M, N, Serial Numbers 2105920 through 210...</td>
<td>14/10/2016</td>
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<tr>
<td>10059783</td>
<td>INTELISAT INFLIGHT LLC</td>
<td>FAA STC ST04370NY</td>
<td>747-400</td>
<td>Installation of Gogo 2KU – Structural and Electrical Provisions</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unarchived and remain applicable to this certificate/approval.</td>
<td>STC Twenty One Ltd Aerial Flight Manual Supplement S21 FMS-0134 Issue/Stc Twenty One Ltd Instructions for Continued Airworthiness S21-TEC-0292 Issue 1. p./p. Appendix A. p./p. Section 3 – Airworthiness Limitations later revisions of the above listed documents approved by EASA-STC Twenty One Ltd.</td>
<td>This design change comprises two installation configurationsApproval is granted for independent installation of these configurations.</td>
<td>28/01/2020</td>
<td>Active</td>
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<td>10059784</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>REV. 1</td>
<td>182S, 182T, T182T</td>
<td>STC-AT-068-RevA - ADAM Cessna Part This minor change is a revision of the already certified STC- AT-068. 10059784. This minor change covers the installation of an additional switch methiopp...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraphs of CS-23 Amendment 7 are applicable to the change: 23.71.23.73.23.74.23.45.23.46.23.51.23.53.23.60.23.65.23.69.23.70.23.72 182S: 23.71.23.73.23.74.23.45.23.46.23.51.23.53.23.60.23.65.23.69.23.70.23.72.</td>
<td>AFM-STC-AT-068, Revision 5, dated 26-FEB-2017, Aircraft Flight Manual Supplement(p.1) p./p.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>19/09/2017</td>
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<td>STC Holder</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td>767-300BCF, 767-400ER</td>
<td>Provision For Dual Portable Electronic Flight Bag (EFB) in a Boeing 787 Series Aircraft.</td>
<td>Doc ref: 209980-AM Flight Engineering Order &amp; Certification ProgramDoc ref: 209980-CM Flight Manual SupplementDoc ref: 209980-COR Compliance ChecklistDoc ref: 209980-A Continued Airworthiness and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/10/2016</td>
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<td>1005984</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SH4477SW</td>
<td>412, 412EP</td>
<td>Installation of a four place utility seat configuration</td>
<td>- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>1210/2016</td>
<td>Active</td>
<td>1005981</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SH4476SW</td>
<td>412, 412EP</td>
<td>Installation of a five place utility seat configuration</td>
<td>- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>1210/2016</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SH4476SW</td>
<td>412, 412EP</td>
<td>Installation of side facing utility seat configuration</td>
<td>- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td></td>
<td>1210/2016</td>
<td>Active</td>
<td>1005981</td>
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<td>FOKKER SERVICES B.V.</td>
<td>ATR 42-500/ATR 72-212 A</td>
<td>Introduction of several cabin layout configurations with LifeSporTmedevac litter system(s) installed.</td>
<td>Compliance Record Report CRR-0269 Issue 1 dated 9/26/2016 AMF Supplement ATR4-STT-AMF-S-001 Revision 1 dated 15/09/2016 STT-AMF-S-002 Revision 1 dated 15/09/2016 Maintenance Recommendations Supplement per Fokker Report SE-482E later revisions of the above listed documents approved by EASA.</td>
<td>Per AMF Supplement ATR4-STT-AMF-S-001 Revision 1 dated 15/09/2016 and ATR4-STT-AMF-S-002 Revision 1 dated 15/09/2016 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td></td>
<td>2120/2016</td>
<td>Active</td>
<td>1005985</td>
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<td>EASA STC no.</td>
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<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10059855</td>
<td>SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT - HELICOPTERS DATAMATIS EUROPE NV - NT SOLUTIONS &amp; SERVICES</td>
<td>REV. 1</td>
<td>G00-00-000</td>
<td>F2000 / F2000EX</td>
<td>Installation of iHELIOM V2+ NUM data recorder/Rev. 1: STC holder name change from I.S.E.I International Services t/o SAFRAN ELECTRONICS &amp; DEFENSE BUSINESS UNIT - HELICOPTERS DATA MANAGEMENT SOLUTIONS &amp; SERVICES.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): Lithium Battery Installations - CRI-F01. The requirements for environmental protection and the associated certified noise and/or emissions levels of...</td>
<td>RFM supplement ref. H11-CRABII G2 H3+ NUM=E-A-01-Rev 01 dated 26/September 2016(+(&lt;)) &gt;or later revisions of the above listed document(s) approved by EASA, STC data list ref. H10-CRABII G2 H3+ NUM=E-A-02 dated 26 September 2020: Instructions for Continued Airworthiness ref. NB CRI-CABII G2 H3+ NUM=E-A-01</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>27/02/2020</td>
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<td>10558602</td>
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<td>10059871</td>
<td>SWS CERTIFICATION SERVICES Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A350-941</td>
<td>Airbus A350-900 for China Airlines – SkyscoutTM Installation Redesign:Rev. 1: Redesign removing need for the CRJ ESF CRI D-2338-001.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- MASTER CONFIGURATION SCHEME: reference 3167-MCS-01 Issue 1 dated 12 September 2017; CERTIFICATION PROGRAMME: reference 3167-CEP-01 at Issue 1 dated 12 August 2017; FLIGHT MANUAL SUPPLEMENT: reference 3144-FMS-01 at Issue 4.</td>
<td>The redesign approved by this STC terminates the need for the CRJ ESF CRI D-2338-001 which will be cancelled. All aircraft fitted with the original STC 10059871 must be retrofitted to the new design as approved through this certificate. The STC 10059871 will be considered revoked prior to installation...</td>
<td>13/09/2017</td>
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<td>10059875</td>
<td>RUAG AG</td>
<td>REV. 1</td>
<td>F2000 / F2000EX</td>
<td>F2000 ADS-B Out: Removed: Rockwell Collins TDR-4D (p/n 622-9210-008 or later) A/C/S/EHS/ELS Transponder Installed: Rockwell Collins TDR-4D (p/n 622-910-501/9/C/S/EHS/ELS) Transponder: A/C/S/EHS/ELS/ADS-B Fail Indicators. A/D-B Inhibit Indicator (Optional).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-ACNS.D.ADS-B 001.010, 005, 025, 026, 031, 035, 040, 045, 055, 060, 070, 080, 085, 090, 105, 110, 115, 120. The...</td>
<td>Flight Manual Supplement Doc. No. FMS-610696, Rev. A, Master Document List Doc. No. MDL-800361, Rev. A, Engineering Order Doc. No. EO-8002496, Rev. B, Instructions for Continued Airworthiness Doc. No. ICA-8003623, Rev. A, or later revisions of the above listed document(s) approved/accepted under this...</td>
<td>This STC is applicable to Falcon 2000 aircraft with: Proline 21 Cockpit-Dual GPS-4000s GPS receivers (832-2189-010)-Dual TDR44 Transponders (832-920-008 or later)- ImplementedDassault Service Bulletin No. 312 (ATC with EnhancedSurveillance) Prior to installation of this change/repair it must be...</td>
<td>02/02/2021</td>
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<td>10558671</td>
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<td>10059876</td>
<td>AIR PLAINS SERVICES, CORPORATION</td>
<td>FAA STC</td>
<td>210R</td>
<td>SA04229W</td>
<td>Installation of McCauley Propeller Model D3A34C40295DFA-8, -10 and associated changes.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>C S Industries Installation Instructions No. 402A, Rev. 1 dated April 1996.</td>
<td>For Models listed above, the installation of McCauley D3A34C40295DFA-8 (propeller is limited to not over 82 inches and not under 78 inches) must be maintained in accordance with the instructions for Continued Airworthiness (ICA) listed in the MDL identified above Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>26/10/2016</td>
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<td>10558676</td>
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<td>10059881</td>
<td>GARMEN INTERNATIONAL, Inc.</td>
<td>FAA STC</td>
<td>210E, F, G, H, K, L, M, N</td>
<td>SA02180E-D</td>
<td>Provisional installation of Garmin GPS3X0 antennas.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>FAA Approved Model List Document No. 055-00383-01, Revision &quot;A&quot;, approved on April 4, 2011, or later FAA Approved revision.</td>
<td>Aircraft modified by this STC must be maintained in accordance with the instructions for Continued Airworthiness (ICA) listed in the MDL identified above Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>26/10/2016</td>
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<td>10558681</td>
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The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment FAR 25 Amendment 25-1-to 25-88 FAR 26.11 and 26.47 Amendment 26-6 CS ACNS subpart D in §25 Appendix J. The requirements in...
List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<td>1005027</td>
<td>ARIES AEROSPACE &amp; MECHANICAL ENGINEERINGS INC.</td>
<td>1</td>
<td></td>
<td>767-300</td>
<td>B767-300 Tapestry Mood Light integration.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Engineering Order Document Number E0_M154001104_MGR_27072016, Rev. 0</td>
<td>Instructions for Continued Airworthiness, Document Number CA_M154001104_MGR_27072016, Rev. 1</td>
<td>02/15/2016</td>
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<td>1005931</td>
<td>AIRBUS HELICOPTERS GERMANY GmbH</td>
<td>REV. 2</td>
<td>MBB-BK117 D-3M</td>
<td>MBB-BK117 D-3</td>
<td>B777-200 SERIES</td>
<td>Installation of BAE Multiple-place Utility Troop Seating System PN119900- (XX), Revision 1 grants approval for the installation of a modified Three-Place Affacing Seating configuration and addresses compliancew with CS 29.1(e), where applicable. Revision 2 grants approval for the installation of an add...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CR D-04 &quot;Passenger emergency exits&quot; Closed at Issue 9, dated 20.08.2020. The requirements for environmental protection and the associated certificate...</td>
<td>Certification System SK-MAC-172-CR-D-1192 Rev D; Certification Program SK-MAC-172-CR-D-1182 Rev F; dated 03.07.2020</td>
<td>13/10/2020</td>
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<td>1005937</td>
<td>JAMCO AMERICA, INC.</td>
<td>FAA STC ST01106E</td>
<td>777-200 SERIES</td>
<td>Installation of new business class seats, installation of previously approved video monitors and relocation of emergency equipment in Boeing Model 777-200 series airplanes.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Certification System SK-MAC-172-CR-D-1192 Rev D; Certification Program SK-MAC-172-CR-D-1182 Rev F; dated 03.07.2020</td>
<td>Compliance Report SK-MAC-172-CR-P-1193-D; dated 18.09.2020.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/11/2016</td>
<td>Active</td>
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<td>10059941</td>
<td>ROSEN SURVIVOR SYSTEMS L.L.C.</td>
<td>FAA STC SA1511M</td>
<td>561 (CITATION II)</td>
<td>Rosen Survivor System – monorail mounted cockpit survivor</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Certification System SK-MAC-172-CR-D-1192 Rev D; Certification Program SK-MAC-172-CR-D-1182 Rev F; dated 03.07.2020</td>
<td>Certification SK-MAC-172-CR-D-1192 Rev D; Certification Program SK-MAC-172-CR-D-1182 Rev F; dated 03.07.2020</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/11/2016</td>
<td>Active</td>
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<td>10059944</td>
<td>AVIONTEC Ltd.</td>
<td>414A</td>
<td></td>
<td>Avionica Upgrade - Installation of Avionica P5 Engineering Printer A00000C - Installation of a NAV/COM 3 System King KX 155A - Replacing Mode 3 XPRD by a Garmin GTX 330 - Replacing Weather Radar System by a GWX 68-Installation L-3 TASI System G-497 System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Certification System SK-MAC-172-CR-D-1192 Rev D; Certification Program SK-MAC-172-CR-D-1182 Rev F; dated 03.07.2020</td>
<td>Certification System SK-MAC-172-CR-D-1192 Rev D; Certification Program SK-MAC-172-CR-D-1182 Rev F; dated 03.07.2020</td>
<td>Certification SK-MAC-172-CR-D-1192 Rev D; Certification Program SK-MAC-172-CR-D-1182 Rev F; dated 03.07.2020</td>
<td>STC applicable for MSN 414-0360, HELIX only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/11/2016</td>
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<td>Limitations/Conditions</td>
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<td>10060000</td>
<td>AGTECH AVIATION SERVICES LIMITED</td>
<td>REV. 3</td>
<td>FLIR system installation on EC135, 737-600, 737-700, 737-800</td>
<td>- Master Document for referenced MSL882-099 Issue 34 dated 15th JAN2024 - Flight Manual Supplement referenced FMS882-019 Issue 19 dated 15th JAN2024 (Applicable to 8737-8)</td>
<td>The applicability of this STC is limited to the pre-configurations of the aircraft identified under the documents ref.DR/882-001 (section 19) applicable to 8737-700-800 or ref. DR/882-002 (section 19) applicable to 8737-8. This STC applicability can be extended to other aircraft MSNs under the terms...</td>
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<td>10060001</td>
<td>FLIGHTSTAR CORPORATION</td>
<td>FAA STC ST04064CH</td>
<td>LEARJET MODEL 45 - Installation of Automatic Dependent Surveillance – Broadcast (ADS-B) Out System.</td>
<td>Flightstar Corporation Master Data List Doc. No. 45-01-002, Revision H dated April 12, 2016, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product 1. The aircraft must be maintained in accordance with...</td>
<td>10/11/2016</td>
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<td>10060002</td>
<td>BOGGI AERONAUTICS S.R.L.</td>
<td>REV. 1</td>
<td>TrakkaBeam® A800 Searchlight System PAL/CP/1280 as described in PhoenixAerospace Limited Report PAL/CEA/01-1 STC transferred to Boggi Aeronautics S.r.l.</td>
<td>- Flight Manual Supplement Document PAL/CFM/1280 Rev. E - Certification Plan PAL/CP/1280 Rev. E - Compliance Record PAL/CP/1280 Rev. E - Instruction for Continued Airworthiness PAL/CA/1280 Rev. E -</td>
<td>The applicability of this STC is limited to the pre-configurations of the aircraft identified under the documents ref.DR/882-001 (section 19) applicable to 8737-700-800 or ref. DR/882-002 (section 19) applicable to 8737-8. This STC applicability can be extended to other aircraft MSNs under the terms...</td>
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<td>10060007</td>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FAA STC ST0297NY</td>
<td>BD-100-1A10 - Installation of ADS-B Out System in the Bombardier BD-100-1A10 (CH300) aircraft</td>
<td>- FAA STC ST0297NY, dated 4th December 2015 - Dassault Aircraft Services Master Data List, ref. doc.CH-300-D000-005, Revision A, dated 4th December 2015 - Airplane Flight manual Supplement AFMS, ref. doc.CH-300-D000-150,Revision IR, dated 4th December 2015 - Certification Plan Document, ref. doc. CH-30</td>
<td>This alteration requires a prerequisite, Installation of Global Positioning System (GPS) with Wide Area Augmentation System (WAAS) Capability – GPS-4000 receivers</td>
<td>09/11/2016</td>
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<td>10060033</td>
<td>ASI INNOVATION</td>
<td>FAA STC ST1027</td>
<td>EC135 T2+ - FLIR system installation on EC135 T2+ Installation of FLIR system with Airliner arm mount (EASSTC 8.M.R.S. 01460) and an operator console.</td>
<td>Supplement Manual de vcl, ASI-APM-0289, Revision 08 dated 2November2016 or later revisions of the above listed documents approved by EASA Certification Program, ASI-CP-0289, Revision 03 dated 19 October 2014 - Compliance Check List, ASI-CCL-0289, Revision 10 dated 2 November 2016</td>
<td>This STC is only valid for eyesafe LASER installations. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the...</td>
<td>10/11/2016</td>
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<td>10660033</td>
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List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<th>Associated Technical Documentation</th>
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<td>DOA AESGLOBAL LIMITED</td>
<td>10060036</td>
<td>A319-111</td>
<td>Installation of Garmin GTN Series GPS/MarkerBeacon Receiver, or:-Option and optional Bendix/King KMR-675 Single Channel Transponder, or:-Option and optional Garmin GDL-69/A XM Weather single Sensor...</td>
<td>Flight Manual Supplement AFMS No.75 Issue 1for later revisions of the above listed documents approved by EASA Master Data List AES-319-300MDL Issue 5</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/04/2021</td>
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<td>VECTOR AEROSPACE HELICOPTERERVICES INC.</td>
<td>10060043</td>
<td>FAA SR02473AK</td>
<td>Installation of the L-3 Communications NGT-9000/RD and NGT-9000/D ADS-BINOUT system.</td>
<td>Master Drawing List, VAHS-1616-1DL, Rev. N C, dated May 11, 2016, or later FAAapproved/accepted revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce roadway effect upon the airworthiness of the product.</td>
<td>02/12/2016</td>
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<td>JET AVIATION AG, BASEL</td>
<td>10060045</td>
<td>ST01691WI</td>
<td>Installation of following antennas:-</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GMBH</td>
<td>10060048</td>
<td>PA-42 (CHEYENNE III) PA-42-720 (CHEYENNE IIIA) PA-42-1000 (CHEYENNE 400LS)</td>
<td>Installation of following antennas:-</td>
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<td>COLUMBIA AVIONICS INC.</td>
<td>10060052</td>
<td>FAA STC ST01691WI</td>
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<td>MT-PROPELLER ENTWICKLUNG GMBH</td>
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<td>FULTSTAN CORPORATION</td>
<td>FAA STC ST0801CH</td>
<td>LEARJET MODEL 45</td>
<td>Installation of Single ULS-1Ew with LPV Monitor Upgrade.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>1. Flightstar Cockpit Master Data List Doc. No. 45-001-001 Revision 2.2 dated January 9, 2016, or later FAA approved revision. 2. Flightstar Supplement, Flightstar Corp. Doc. No. 45-818-001 Revision NC, approved February 8, 2016,&lt;sup&gt;[1]&lt;/sup&gt; or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>14/11/2016</td>
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<td>SEGELFLUGGRU e.V.</td>
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<td>HK 36 TTC</td>
<td>Increase MTOW from 770kg to 800kg</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. This certificate approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>-Erläuterung für das Flughandbuch des Motorschleppers HK 36 TTC Series Nr. 36525 Betrieb mit mehr als 770 kg bis maximal 800 kg Fluggewicht.</td>
<td>This change is limited to Serial number 36525/Phil to installation of this design change. It must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/11/2016</td>
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<td>AERODATA AG</td>
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<td>B200GT 200</td>
<td>Suppression of ADS-B OUT</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-23 Amdt 423.1; 23.1342, 23.3(a), 23.135(a) and (b), 23.256c(c), 23.1365, 23.305; 23.1367, 23.403, 23.1501, 23.773 (a), 23.1509, 23.1310(b)(1),(2) and (d), 23.154(b).</td>
<td>Installation, Operation and Maintenance according to the Master Drawing List document number MDL-A/ C-94 Rev. 2 dated 11th November 2016 or later revisions of the above listed documents approved by EASA.</td>
<td>Applicability limited to B200/ B200GT equipped with Rockwell Collins ProLine 21 avionics/Rockwell Collins Dual TDR94/94D transponder installation versions with/without ADS-B (Automatic Dependent Surveillance) and/or ADS-B/Dependent Automatic Surveillance-Broadcast) capability.</td>
<td>14/11/2016</td>
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<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td>AS 350 D AS 350 Bi AS 350 B1 AS 350 B2 AS 350 B3 AS 350 BAAS 350 BB</td>
<td>Installation of Video Transmission System on AS350 series rotorcrafts. This includes a cabin rack, a cabin display mount and two antennas located on landing skids.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>PP-079-AMF-03 Supplementation to Aircraft Flight Manual. Issue 03, dated 14/10/2016 or later revisions of the above listed documents approved by EASA. PP-079-CA-04, Supplement to Instruction for C.A., initial issue dated 08/07/2016. No ALS is identified for this change. PP-079-MLD-04, Master Data List, issue...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>15/11/2016</td>
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<td>TRUATLANTIC MFG., LLC</td>
<td>FAA STC SA0796AT</td>
<td>206/ 207/ 210</td>
<td>Installation of a dual isolator forward engine mount system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Aero Modifications &amp; Consulting, Inc. “21 004021 Installation/Instructions” Rev. C dated June 23&lt;sup&gt;[1]&lt;/sup&gt;,(1) &gt; 1995, and (2) 21040550 Master Drawing List Rev. C dated June 23, 1995, or later FAA approval revision of either document.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>16/11/2016</td>
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<td>10060017</td>
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<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>BD-100-1A10</td>
<td>Astrar 350 Satellite Communication System With Wireless Local Avionics Network (WLAN) including 3G/4G Network Functionality On Ground Only.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment EWS: Applied i a.w. (generic) SOC/IRI H-01 and ACUS-2TA-CS 25.1, CS 25.29, CS 25.251, CS 25.301, CS 25.303, CS 25.305, C...</td>
<td>Doc ref: 2087703-EO Engineering Order &amp; Certification Program; Doc ref: 2087703-AM-4 Flight Manual Supplement; Doc ref: 2087703-COR-2 Compliance Checklist; Doc ref: 2087703- CA-2 Continued Airworthiness later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>15/11/2016</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>SIRIUM AEROTECH S.L.</td>
<td>10/06/2016</td>
<td>Active</td>
<td>10060002</td>
<td>MYSTERE FALCON 900 (MF900)</td>
<td>GPS upgrade for Falcon 900.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS25, Amendment 17 paragraph 25.1316 and 25.1317. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>MASTER DATA LST, Ref. SA16-545-MLD, Rev. 1, Oct 5th 2014, applicable to the abovementioned STC revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/02/2016</td>
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<td>AIRBUS HELICOPTERS, Inc.</td>
<td>10/06/2016</td>
<td>Active</td>
<td>10060004</td>
<td>EC 130 T2</td>
<td>Installation of Garmin G500H Electronic Flight Instrument System (EFIS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- American Eurocopter Drawing List 130A8-80109, Rev. D, dtd. 14 March 2013; Rotorcraft Flight Manual Supplement 130FMS-506, Rev. R1, dtd. 27 March 2013; Instructions for Continued Airworthiness 130FMS-600, Rev. R3, dtd. 23 February 2013; later revisions of the above listed documents approved by EASA.</td>
<td>- This G500H EFIS installation approves the Garmin GDU 620 display (left hand PFD and MAP, GFRS 77H ARP, GMU 44 Magnetometer, GDC 74HDGC and GTP 59 GAT only. This STC only approves the installation of Garmin GNS540W/500W series GPS/ NAV/COM, GMA340H audio panel, and GTX430 transponder to the G-100.</td>
<td>11/02/2016</td>
<td>Active</td>
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<td>AERODATA AG</td>
<td>10/06/2016</td>
<td>Active</td>
<td>10060009</td>
<td>1900D</td>
<td>Installation of 230VAC Power Supply.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements. The following paragraph CS-23, Amendment 4 paragraphs apply: 25.1323(b).3(c), 25.133(c), 25.3, 25.305(a), 25.306(a), 25.306(b), 25.307(a) and (b), 25.341(a) and (c), 25.65</td>
<td>Installation, Operation and Maintenance according to the Master Drawing list number MDL-AT-C-888 Rev. 4, dtd. 11th November 2016; later revisions of the above listed documents approved by EASA.</td>
<td>Applicable to Beechcraft 1900D with installed Rockwell Collins Pro Line Collins with an AD-AFIS-020 flight inspection system installed/EASA STC 1005031 Revision 1 (or latest EASA approved revisions) is applicable to this installation. Prior to installation of the design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/02/2016</td>
<td>Active</td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>10/06/2016</td>
<td>Active</td>
<td>10060010</td>
<td>A340-300</td>
<td>Interior reconfiguration of Lufthansa aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Summary, Northwest Aerospace Technologies, Inc. document No. N545-D0, Revision 1, dtd. 14 November 2018; Instructions for Continued Airworthiness, Northwest Aerospace Technologies, Inc. document No. N545-D70, Initial Release, dated 28 April 2010; later revisions of the above lists.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/02/2016</td>
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<td>SABENA TECHNICS BGC</td>
<td>10/06/2016</td>
<td>Active</td>
<td>10060011</td>
<td>REV. 1</td>
<td>A340-642, A340-643 A340-541, A340-542 A340-311, A340-312, A340-313 A340-211, A340-212, A340-213</td>
<td>Assessment of AC Compatibility T-PED on A340 Aircraft Revision 01 : Revision of the approval following the use by the possessant of the EASA CM-ES-003 Issue 1 (Guidance to Certify an Aircraft T-PED tolerant), and modification of the associated installation/conditions information.</td>
<td>- Instructions for Continued Airworthiness, Aerocore document No. 0006-01-A-API 1051-03; - Aircraft Flight Manual Supplement, Aerocore document No. 0006-01-A-APA 501-03; - later revisions of the above listed documents approved by EASA Master Data List; Aerocore document No. 0006-01-A-APA 501-04.</td>
<td>1) This STC does not constitute installation approval of aeropane/airframe, it only provides data that may partially support their installations or activations. Even if the airplane is not fitted with an active in-flight communication system, this system is installed/airframe-approved in this configuration.</td>
<td>12/06/2016</td>
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<tr>
<td>HELICOPTER AIR TRANSPORT GmbH</td>
<td>10/06/2016</td>
<td>Active</td>
<td>10060012</td>
<td>EC135 T2+</td>
<td>Miscellaneous changes for CFV operation/This change covers the conversion of the configuration of rotorcraft serial number 0374 from configuration used by the previous operator to &quot;Christophorus Flugbetreuungen&quot; (CFV) delivery configuration of rotorcraft serial number 0600 to meet CFV fleet communion.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements. The following paragraphs of CS 27 Amendment 3 are applicable to the change: 27.1; 27.301; 27.303; 27.305; 27.307; 27.361; 27.600; 27.602; 27.610; 27.771; 27.1301; 27.1309; 2.</td>
<td>EO 85-011, Engineering Order, Revision 6, dated 10 Oct 2016, including referenced documents.</td>
<td>Limited to rotorcraft serial number 0374.Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/02/2016</td>
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<td>10060139 T6 Klein Aviation Inc.</td>
<td>REV 1</td>
<td>FAA SA098900AC</td>
<td>T206H</td>
<td>Increase the maximum ramp weight to 1726 kg (3,806 pounds) and maximum takeoff weight to 1718 kg (3,789 pounds).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>- Aircraft Master Drawing List 2900 Rev. A dated May 17, 2016. Alamo Aerospace Flight Manual Supplement number AFMS 55-196-5 Rev. (Related June 13, 2016) or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/94/CF (or subsequent revisions thereof).</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/12/2016</td>
<td>Active</td>
<td>10060139</td>
</tr>
<tr>
<td>10060149 328 Support Services GmbH</td>
<td>REV 1</td>
<td></td>
<td>DORNIER 328-100</td>
<td>Empty Cabin Layout, Project No. 089.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List MXL098201 Iss. B dated 18.11.2016; Declaration of Compliance DC-12809920001 Iss. B dated 18.11.2016; Compliance Statements CS-12809920001 Iss. A dated 25.10.2016; A/C Flight Manual Supplement AFMS51280992001 Iss. B dated 15.11.2016 or later revisions of the above listed documents.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/09/2018</td>
<td>Active</td>
<td>10060149</td>
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<tr>
<td>10060172 Societe Air France S.A.</td>
<td></td>
<td></td>
<td>777-200</td>
<td>AZ – Cabin rearrangement; The modification consists in a cabin arrangement with 293 passengers including GEVEN Economic and Premium Economic Class seats installation and STELIA Business class seats modification.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Change Approval Sheet CAS-B777-1S-014 or 0-Instruction for Continued Airworthiness, ref. ICA- B777-1S-014-001 rev0 or later revisions of the above listed documents approved by EASA.</td>
<td>Monitors and components of the IFE System including In-Seat Power Supply System are treated as pure items of mass and are subject to a change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/11/2016</td>
<td>Active</td>
<td>10060172</td>
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<tr>
<td>10060173 Bombardier Inc.</td>
<td></td>
<td></td>
<td>BD-700-1A10</td>
<td>EASA validation of Transport Canada Supplemental Type Certificate SA03-77 at Issue 1 Installation, Vacuum Waste System Vent</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Modification to be completed in accordance with Aero Consulting Services Modification Data Summary DS89700 Revision NC/Mandatory Maintenance Manual Supplement BAS67060000 Revision NC; or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of the Canadian Bilateral Agreement.</td>
<td>Mandatory aircraft maintenance requirements (airworthiness or emissions levels) and required aircraft equipment as per Mandatary Maintenance Manual Supplement BAS67060000 Revision NC, or later revisions approved by EASA in accordance with the Technical Implementation Procedures of the Canadian Bilateral Agreement.</td>
<td>23/11/2016</td>
<td>Active</td>
<td>10060173</td>
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<tr>
<td>10060188 Micro Aerodynamics Inc.</td>
<td></td>
<td>FAA STC SA098900SE</td>
<td>114A/ 114B/ 114TC</td>
<td>Installation of vortex generators on the wings, horizontal tail surfaces and vertical fin.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Micro Aero Dynamics Drawing Package MA2128, Revision IR, dated November 1, 2000, and Installation Manual MA2128 Revision IR, dated December 1, 2000; or later FAA approved revisions.</td>
<td>If more than five vortex generators are damaged or missing (&lt; &gt; the airplane is not airworthy. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/11/2016</td>
<td>Active</td>
<td>10060188</td>
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<td>STC Holder</td>
<td>FAA STC/STC ref.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>DASSAULT FALCON JET</td>
<td>STO1408W-I-D</td>
<td>Installation of PRIMUS EPIC Control Display System for Retrotif (CDS/Rijn</td>
<td>The Certification Basis for the original product remains applicable to certificate</td>
<td>Document List, No. 060509005, Revision U, dated 14 January 2014; Dassault Aviation Inc.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other subsequently installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/02/2020</td>
<td>Active</td>
<td>10060212</td>
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<td>CORPORATION</td>
<td></td>
<td>Dassault Aviation Mystere-Falcon 200 Airplanes originally equipped with</td>
<td>approval, except where amended by additional or later amendments if indicated on FAA STC</td>
<td>FAA Approved Airplane Flight Manual Supplement, No. 060509008, Revision J, dated 14 January 2014;</td>
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<td>the Phase 5 update with Localizer Performance with Vertical Guidance (LPV)</td>
<td>certification and noise and/or emissions levels of the...</td>
<td>060509008, Revision J, dated 14 January 2014; Dassault Aviation Inc., FAA Approved Airplane Flight</td>
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<td>approv...</td>
<td>approval, except where amended by additional or later amendments if indicated on FAA STC</td>
<td>Manual Supplement, No. EDOC-000016750...</td>
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<td>MARSHALL OF CAMBRIDGE</td>
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<td>Auxiliary Fuel Management System Provision (AFMS). This modification</td>
<td>The Certification Basis for the original product remains applicable to certificate</td>
<td>- Installation must be in accordance with EO N° MUI-00-0010; Issue 9 - Engineering Order for</td>
<td>AFMS must not be installed on aircraft with CAT operations. The aircraft must not operate at a</td>
<td>25/11/2016</td>
<td>Active</td>
<td>10060222</td>
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<td>AEROSPACELIMITED</td>
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<td>provides the mechanical and electrical systems provisions within the existing</td>
<td>approval, except where amended by additional or later amendments if indicated on FAA STC</td>
<td>Introduction of the Q400 Auxiliary Fuel Management System: Continued Airworthiness must be</td>
<td>range greater than 1 hour's flyingtime from an adequate aerodrome at the one-engine</td>
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<td>DHC-8-401, DHC-8-402</td>
<td>STO1408W-I-D. The requirements for environmental protection and the associated</td>
<td>accordance with ICA/Q400(1995)/15/0667, Issue 2 and Airplane Maintenance Manual MTM2096 Part 1 &amp;</td>
<td>inoperative cruise speed, in standard atmospheric conditions and still as Activation of the AFMS and conditions for operation...</td>
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<td>certification and noise and/or emissions levels of the...</td>
<td>MTM2033 Part 2, Rev.1-E...</td>
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<td>approval, except where amended by additional or later amendments if indicated on FAA STC</td>
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<td>TELEDYNE CONTROLS, LLC</td>
<td>STO1823LA</td>
<td>Wireless GroundLink®-Quick Access Recorder System installation. This</td>
<td>The Certification Basis for the original product remains applicable to certificate</td>
<td>- Master Data List, ref. MOL-2443800-A340 Revision D, dated 24 September 2013; Instructions</td>
<td>None</td>
<td>25/11/2016</td>
<td>Active</td>
<td>10060224</td>
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<td>modification installs the Teledyne Control, Wireless GroundLink-Quick</td>
<td>approval, except where amended by additional or later amendments if indicated on FAA STC</td>
<td>for Continued Airworthiness, ref. ICA2443800_A4_1, dated 19 May 2015; for later revisions of the</td>
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<td>Access Recorder (WDAR), P/N 2243800-xx, together with the enhanced airborne</td>
<td>STO1408W-I-D. The requirements for environmental protection and the associated</td>
<td>above listed documents approved by EASA in accordance with ICA/Q400(1995)/15/0667, Issue 2 and</td>
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<td>Data Loader (eADL), P/N 2235300-xx, Data Loader (WQAD), P/N 2235300-xx</td>
<td>certification and noise and/or emissions levels of the...</td>
<td>Airplane Maintenance Manual MTM2096 Part 1 &amp; MTM2033 Part 2, Rev.1-E...</td>
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<td>approval, except where amended by additional or later amendments if indicated on FAA STC</td>
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<tr>
<td>TRUATLANTIC MPD, LLC</td>
<td>STO-00252SAT</td>
<td>Installation of Engine Mount Brackets.</td>
<td>The Certification Basis (CS) for the original product remains applicable to this certificate/</td>
<td>- Instructions for Continued Airworthiness, dated April 5, 2002, Revision &quot;R&quot; for later</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any</td>
<td>25/11/2016</td>
<td>Active</td>
<td>10060225</td>
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<td>approval, except where amended by additional or later amendments if indicated on FAA STC</td>
<td>revisions of the above listed documents approved by EASA in accordance with ICA/Q400(1995)/</td>
<td>otherwise installed design change and/or repair will introduce no adverse effect upon the</td>
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<tr>
<td>ICELANDAIR EHF</td>
<td>STO-00252SAT</td>
<td>Communications - Data Transmission &lt; +ACARS SYSTEM INSTALLATION</td>
<td>The Certification Basis for the original product remains applicable to certificate</td>
<td></td>
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<td>28/11/2016</td>
<td>Active</td>
<td>10060231</td>
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<td>approval, except where amended by additional or later amendments if indicated on FAA STC</td>
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For more information, visit [easa.europa.eu](https://easa.europa.eu).
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<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Movies</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>10080239</td>
<td>2 EXCEL DESIGN LIMITED</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>B300, B300C, B200CT, B200GT, B200T, B200C, B200CT, B200, 200C, 200CT, 200T</td>
<td>Approval of the Introduction of Standalone GPS Antenna on Beachcraft/King Air 200 and 200C all variants, under Leading Edge Modification nr2XL030, issue 1, dated, 21 October 2016.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Instructions with Leading Edge Master Drawing nr2XL030 LIF78(1), Issue 1, dated, 21 October 2016 and Installation Instructions ref. nr. 2XL030 LIF494(1), Issue 1.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/11/2016</td>
<td>Active</td>
<td>10080239</td>
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<tr>
<td>10060240</td>
<td>ALTITUDE AEROSPACE INTERIORS Ltd.</td>
<td></td>
<td>787-9</td>
<td>Cabin:&lt;&lt;p&gt;&gt; A Zone Bar; Perch Cushion Installation. The modification entails the removal of &quot;No Sit&quot; devices from perch/beam replacement with fitted cushions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Summary report ref DP: 00540 C ICA, CMM 25-20-17 (El 010107) Rev C MDL ref MDL 22001142, Cor later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/11/2016</td>
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<td>10060240</td>
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<tr>
<td>10080244</td>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA ST04270AT-D</td>
<td>G280</td>
<td>Installation of Aerial View System (AVS) Video Tail Camera.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Index List GA0509001, Revision M, dated 30 July 2014.</td>
<td>The following EASA STC is a required part of this STC. 10080248 Installation of an Executive Cabin Interior. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/11/2016</td>
<td>Active</td>
<td>10080244</td>
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<tr>
<td>10080249</td>
<td>SWS CERTIFICATION SERVICES Ltd.</td>
<td>REV. 1 CAA UK RESPONSIBIL ITY</td>
<td>777-300ER</td>
<td>Air Austral 777-300ER Skycouch Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>SWSS Certification Services Ltd Master Configuration Scheme No 3176-MCS-01 Issue 7 for later revisions of the above listed documents approved by EASA.</td>
<td>Skyouch can only be used as a sleeping area or as an accommodation area, as detailed in SWS Certification Services Ltd Design Report/176-DR-01, during in-flight conditions.</td>
<td>12/12/2018</td>
<td>Active</td>
<td>10080249</td>
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<tr>
<td>10080254</td>
<td>SABENA TECHNICS BOD</td>
<td>A330-323</td>
<td>Air Caraibes - Ultra High Comfort Attendant Seats (UHAS) installation. The STC installs 4 UHAS together with the G2 seat configuration and modifications to Flight Entertainment system already installed.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Modification Approval Sheet Ref. MAS 5-S-25-455 Issue 3 dated 26 November 2016.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/11/2016</td>
<td>Active</td>
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<td>10080260</td>
<td>AVIONTEC Ltd.</td>
<td>PA-28-181 (ARCHER III)</td>
<td>LPI/ Approach Certification</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- To be operated according to AFM Supplement No. P11543-011, Rev 2, dated 14 November 2018.</td>
<td>Limited to the S/N: 2843095 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/11/2016</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10060272</td>
<td>NSE</td>
<td>C16/14LT/</td>
<td>Evolution of the UMS Brits Saver 2 from amendment 7 to 8.</td>
<td>Certification Plan.</td>
<td>P6CERT-01297-02-EN Rev 00Definition File/0027-02-EN Rev 19Plan for Software Aspect of Certification/PSAC-0029-02-EN Rev 04Software Accomplishment Summary SAAS-0029-00-EN Rev 10Software Configuration Index SCI-0030-02-EN Rev Exp 10STC Document List SDL-BS2-A3332 Rev 056 later revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a road hazard upon the airworthiness of the product.</td>
<td>30/12/2016</td>
<td>Active</td>
<td>10060272</td>
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<td>10060283</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A340-313</td>
<td>A340 EDW Business Class Installation</td>
<td>- Supplemental Type Definition Document, Ref. STD2000-1820Issue 1 Rev0-ICA Rev.</td>
<td>A-00-16/72-ICA Issue 1 Rev 1-Compliance Record List: Ref. CERL01-16/20M Issue 1 Rev 0r later revisions of the above listed documents approved by EASA.</td>
<td>This approval applies to those A330-313 aircraft in which the following modifications have been installed: Changes in the business class area as defined in Airbus SBA340-25-2493 Installation Guide, a non-electrical floor path marking system, by HT Engineering Bulletin A340-EB3.</td>
<td>01/12/2016</td>
<td>Active</td>
<td>10060283</td>
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<td>10060285</td>
<td>BABCOCK MISSION CRITICAL SERVICES DE SIG AND COMPLETIONS LIMITED</td>
<td>EC135 T3</td>
<td>BABCOCK/MOD/125 EC135</td>
<td>- Flight Manual Supplement FMS/125-002, Issue 3, dated 25 Nov 2016</td>
<td>FAA approved revision of the above listed documents approved by EASA.</td>
<td>This change is applicable to the helicopter with CPDS. This is a series change applicable to EC135 T3 aircraft equipped with Airbus helicopters NVIS optional lighting and equipment as defined in drawing BDI-125-100. Any deviation from the cockpit or cabin configuration specified in this STC may not be installed.</td>
<td>01/12/2016</td>
<td>Active</td>
<td>10060285</td>
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<td>10060296</td>
<td>REB TECHNOLOGIES Inc/UK/REB/TECH</td>
<td>R44 ii</td>
<td>FAA STC SR09489RC</td>
<td>R44 installation of Night Vision Goggle Compatible Lighting System.</td>
<td>REB Technologies, Inc, Master Drawing List 98020044MDL Rev. P, dated October 27, 2012; or later FAA approved revision. EASA Flight Manual Supplement (RFMS) 98020044RFMS, Rev. A-dated March 30, 2012, or later FAA approved revision; or 98020044SRM, Rev IR, dated October 31, 2012 or later FAA approved revision; or 98020063SRM, Rev IR, dated November 20, 2012 or later FAA approved revision; or 98020063SRM, Rev IR, dated November 20, 2012 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a road hazard upon the airworthiness of the product.</td>
<td>10/02/2017</td>
<td>Active</td>
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<td>10060308</td>
<td>SABENA TECHNICS BGC</td>
<td>A330-343</td>
<td>Cabin Reconfiguration with IFE/SPSS</td>
<td>- Master Data List, ref 4933-01-A-ML-000-Instructions for Continued Airworthiness, ref 4933-01-A-ICA-R01; - Aircraft Flight Manual Supplement, ref 4933-01-A-APMS-R00 later revisions of the above listed documents approved by EASA.</td>
<td>It is required the installation of Non-Electrical Floor Path Markingsystem, STC EASA 10014507 Rev 2. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a road hazard upon the airworthiness of the product.</td>
<td>02/12/2016</td>
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<td>EASA STC no.</td>
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<td>10060315</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>REV. 1</td>
<td>FAA STD SR20508SE</td>
<td>MBB-BK117 D-3 MBB-BK117 D-2</td>
<td>Installation of D117-1028-D111-013 Bearpaw Kits Revision 1 adds the MBB-BK117 D-3 model.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>1. Installation Instructions IN-D117-1028, Revision D, dated 25 January 2023. Master Document List (MDL) D117-11028, Revision D, dated 25 January 2023. Instructions for Continued Airworthiness (ICA) ICA-D117-1028, Revision 3, dated 25 January 2023 or later revisions of the above listed document.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/vibration effect upon the airworthiness of the product.</td>
<td>20/06/2023</td>
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<td>10060315</td>
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<td>10060318</td>
<td>E.A.D AEROSPACETECHNICAL</td>
<td>REV. 1</td>
<td>A321-251N-253N-271N-272 N A321-2111-2121-2131-2311-232</td>
<td>A321-1511-1121-131 A320-2111-2121-2141-2151-216 A319-1211-1221-123 A319-1111-1121-1141-115</td>
<td>Installation of antenna structural provisions for AOGC System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>- Master Document List, EAD Aerospace document No. 01512-EAS-A-MDL, Revision D, dated 21 September 2017. Instructions for Continued Airworthiness, EAD Aerospace document No. 01512-EAS-A-ICA-01, Revision D, dated 20 September 2017 or later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/vibration effect upon the airworthiness of the product.</td>
<td>29/09/2017</td>
<td>Active</td>
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<td>10060319</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>1900D</td>
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<td>Installation of Altitude Select function for B787000 aircraft in compliance with Aerotec Concept Master Data List AE16151/MDL, Issue 0, dated 20 October 2016.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>Aircraft Flight Manual Supplement, Aerotec Concept document AE16151/AFMS, Issue 0, dated 20 October 2016 or later revisions of the above listed documents approved by EASA for EASA Instructions for Continued Airworthiness, Aerotec Concept document AE16151/ICA, at issue 0, dated October 10, 2018.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/vibration effect upon the airworthiness of the product.</td>
<td>05/12/2016</td>
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<td>10060323</td>
<td>LUFTHansa TECHNICAL services gmbh</td>
<td>A380-800</td>
<td></td>
<td>Installation of Data Broadband Transceiver System - Activation of PACON-Board Mobile Telephony System (OMTS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>Compliance Record List Ref. CRL44-16/03/1 issue 1 dated 21 NOV 2016. Airplane Flight Manual Supplement Ref. A-00-16/69-4AFM Issue 1 dated 15 SEP 2016. Master Minimum Equipment List Supplement Ref. A-00-16/69-MMEL Rev. 0 dated 15 SEP 2016 Instruction for Continued Airworthiness Ref. A-00-16/69/ICA issue 1.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/vibration effect upon the airworthiness of the product.</td>
<td>05/12/2016</td>
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<td>10060369</td>
<td>CESSNA ZURICH CITATION SERVICE CENT/GBH</td>
<td>REV. 1</td>
<td>BAE.125 SERIES 800A</td>
<td>Installation of Dual Universal FMS-1Fw (SBAS) with TAWS Class A for LNAV/approach.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-ACN1 Initial Issue. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>- Master Document List, Design Engineering Zurich document No. MDL-116 Revision A, dated 7 December 2016. - Instructions for Continued Airworthiness, Design Engineering Zurich document No. ICA-116, Revision A, dated 1 December 2016. - Aircraft Flight Manual Supplement, Design Engineering Zurich document...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/vibration effect upon the airworthiness of the product.</td>
<td>08/12/2016</td>
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<td>STC Holder</td>
<td>Rev.</td>
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<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>THALES AVIONICS Inc.</td>
<td>REV. 2</td>
<td>FAA ST03414NY</td>
<td>A321-231, A321-232, A321-211, A321-212, A321-213, A321-111, A321-112, A322-231, A322-111/121-131/211-212</td>
<td>Installation of a Thales In-Flight Entertainment System Rev. 1. Transfer of the STC from Delta Engineering to Thales Avionics, Inc. Rev. 2. Change of Delta Engineering MD/ICA to Thales MD/ICA.</td>
<td>The Certification Basis for the original product as amended by thresholding additional or alternative airworthiness requirements - The following paragraph(s) at a later amendment: CS25.853 at Amtd 13, CS25.854 and Amtd 4 and CS25.981 at Amtd 13. The requirements for environmental protection and the ...</td>
<td>- Thales In-Flight Entertainment System Rev A dated 15 April 2021. Thales Master Drawing List 501550-MDL Revision C dated 15 April 2021. Instructions for Continued Airworthiness, 501550-ICA Revision A dated 13 April 2021 or later revisions of the above listed documents approved by EASA in accordance with EASA. Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/01/2021</td>
<td>Active</td>
<td>10063071</td>
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<tr>
<td>DELTA G DESIGN Inc.</td>
<td>FAA STC</td>
<td>ST1112055C</td>
<td>737-900ER</td>
<td>Installation of a Rockwell Collins Tailwind SSR Antenna on a Boeing 737-900ER</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Delta G Design Inc. FAA STC Supplement DGD-CC550-AFM-01/Revision IRDelta G Design Inc. Instructions for Continued Airworthiness No.DGD-CC550-IC-01 Rev Aor later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/48/EC (or subsequent revisions). Note that the ICA report DGD-C550-IC-01 Rev A contains additional mandatory airworthiness inspections. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/12/2016</td>
<td>Active</td>
<td>10063082</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC</td>
<td>SR01186AT</td>
<td>205A-1, 212, 412, 412EP</td>
<td>Installation of replacement fuel cap or locking fuel cap ontoICARoC Helicopter Models 205A-1, 212, 412, 412EP.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Aeronautical Accessories, Inc. report no. AA6134. &quot;Replacement Fuel Cap Kit, Locking Fuel Cap Kit, Replacement Fuel Cap Installation Instructions,&quot; N/No. dated 22 August 1998 or later FAA approved revision.</td>
<td>The approval of this change in type design is restricted to helicopters identified above only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/12/2016</td>
<td>Active</td>
<td>10063063</td>
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<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 1</td>
<td>A330-343</td>
<td>Cabin Reconfiguration (Intrados Aviation Management Rev 1, Transfer of STC from SR Technic Switzerland to groWING Engineering Partners GmbH)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Flight Manual Supplement: S.OADE.000454-AFM Rev 03Master Document List: S.OADE.000454-MDL Rev 01Compliance Check List: S.OADE.000454-CCS Rev 01Instructions for Continued Airworthiness: S.OADE.000454-ICA Rev 00or later revisions of the above listed documents approved by EASA.</td>
<td>The approval of this change in type design must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/04/2021</td>
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<td>COMTECH GLOBAL AVIATION SERVICES S.</td>
<td>REV. 1</td>
<td>737-800</td>
<td>Introduction of 174 passenger seating configuration</td>
<td>The Certification Basis for the original product as amended by thresholding additional or alternative airworthiness requirements: CS 25.23(a)(x)(1), CS 25.25(a)(x), CS 25.27, CS 25.29, CS 25.301(a)(b), CS 25.303, CS 25.305(a)(x), CS 25.307(a), CS 25.309(a), CS 25.315(a), CS 25.601, CS 25.602, CS 25.605, CS 25.607...</td>
<td>- Master Data List, Ref. CTE-MDL0144-737, Rev. &quot;-&quot;, dated Dec 2016. Instructions for Continued Airworthiness, Ref. CTE-TP0144-01, dated Nov 17, 2016or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/06/2019</td>
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<td>10063096</td>
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<td>BLR AEROSPACE, L.L.C.</td>
<td>REV. 4</td>
<td>FAA SR00659RC</td>
<td>AS350 D0 AS350 B, B1, B2, B3 AS350 B C AS350 D0 B AS350 D1 AS350 D2</td>
<td>Installation of BLR FASTRIN® System Rev. 1. Transfer of STC from JASPP Engineering Services LLC to BLR Aerospace LLC. Rev. 2. Update of STC holder's address. Rev. 3. Addition of RFMS-350FF-20, to the AML, dealing with compatibility/.setHeight/Aircraft Increased Take-off Power (Mod. 07.5102 RFMS/Spline)...</td>
<td>The Type and OIS Certification Basess(CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval w...</td>
<td>See &quot;FAA Approved Model List for STC No. SR00659RC Revision IR, dated 28 February 2022 for applicable Installation Manuals, Rotorcraft Flight Manuals Supplements and Instructions for Continued Airworthiness For EASA Validation of 03 AS 350 B3 with Mod. 07.5102, RFMS-350FF-20, Rev2, dated 2023/03/02 appl...</td>
<td>This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List SR00659RC. Rev. IR, dated 2 February 2022. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design changes and/or repairs will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/11/2023</td>
<td>Active</td>
<td>10063098</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td>FAA STC 03817AT</td>
<td>B717-200 SERIES</td>
<td>Installation of a Thales-KANNAD Emergency Locator Transmitter in B717-200 Series Aircraft.</td>
<td>- Avionics Support Group, Inc. Master Data List 29035-MDL Rev. A, dated 08 October 2009, or later. FAA approved Airplane Flight Manual Supplement for Avionics Support Group 29055-FMS dated 29 October 2009 or later. FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that: the interrelationship between this design change and any other previously installed design change and/or repair will introduce an adverse effect upon the airworthiness of the product.</td>
<td>09/12/2016</td>
<td>Active</td>
<td>10060403</td>
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<td>INTERNATIONAL ENGINEERING SISTANCE (IEA)</td>
<td>REV. 2</td>
<td>EC135 T2+, EC135 T3</td>
<td>TIR/VH station installation in cabin. Rev. 2: New cabin layout with Airbus Helicopter passenger seats added.</td>
<td>- Master Document List SCA-00740-MDL-R02 Rev. 5, dated 29 October 2016.</td>
<td>Can only be installed with prior installation of EASA STC 10055845. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a roadw Alexander Effect upon the airworthiness of the product.</td>
<td>15/11/2017</td>
<td>Active</td>
<td>10060425</td>
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<td>PMV ENGINEERING</td>
<td>737-900, 737-900ER 737-600, 737-700, 737-800</td>
<td>TPED tolerance on B737 NG</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List SCA-00740-MDL-R02 Rev. 5, dated 29 October 2016.</td>
<td>Per Instructions for Continued Airworthiness document SCA-00740-ICA-R00, dated 29 September 2016. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change is acceptable.</td>
<td>12/12/2016</td>
<td>Active</td>
<td>10060428</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC 03940NY</td>
<td>777-300 777-300ER</td>
<td>Installation of Panasonic 3000 IFE System.</td>
<td>Master Data List, Ref. 0143-06439-2182 Rev. U, dated 09/12/2016.</td>
<td>The EASA STC covers only the configuration no. 11 of the MDL, Ref. 0143-06439-2182 Rev. U. The prior installation of the EASA STC 10055845 is required. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change is acceptable.</td>
<td>13/12/2016</td>
<td>Active</td>
<td>10060430</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC 03967NY-D</td>
<td>777-200, 777-200LR</td>
<td>Installation of Panasonic 3000 IFE System.</td>
<td>- Master Data List, Ref. 0143-07671-2288 Rev. U, dated 11/12/2016.</td>
<td>This STC includes an AMOC to AD2006-11-13. Prior to installation of this design change it must be determined that: the interrelationship between this design change and any other previously installed design change and/or repair will introduce a road side effect upon the airworthiness of the product.</td>
<td>12/12/2016</td>
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<td>Associated Technical Documentation</td>
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<td>TENENCIA Ltd</td>
<td>TENENCIA Ltd</td>
<td>REVT.1</td>
<td>10060441</td>
<td>A330-322/324-324</td>
<td>WIRELESS IN-FLIGHT ENTERTAINMENT (Wi-JFE) SYSTEM.</td>
<td>TENENCIA LTD Flight Manual Supplement No. TEN 797-47 issue (&gt; &gt;) &gt;r later revisions of the above listed documents approved by EASA.</td>
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<td>T.P.S. S.P.A.</td>
<td>T.P.S. S.P.A.</td>
<td>REVT.1</td>
<td>10060444</td>
<td>AW189</td>
<td>Installation of Medical Oxygen System in accordance with Design ChangeTPS189-001-00. Rev 1. Change of the approval holder name from TPS AEROSPACEENGINEERING S.r.l. to T.P.S. S.P.A.</td>
<td>The Type and OSI Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
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<td>EMBRAER ENGINEERING &amp; TECHNOLOGY INTL USA, INC.</td>
<td>T.P.S. S.P.A.</td>
<td>REVT.1</td>
<td>10060447</td>
<td>FAA STC SA03903CE</td>
<td>Installation of Medical Evacuation Configuration (MEDEVAC) (This STC was originally issued as a validation of Brazilian CST (STC No. 2012/325-15) (11 February 2016). STC transferred to United States based Design Approval Holder (DAAH) on 13 May 2019.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
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<td>SKYTRAC SYSTEMS Ltd.</td>
<td>SKYTRAC SYSTEMS Ltd.</td>
<td>REVT.1</td>
<td>10060483</td>
<td>LEARJET MODEL 45</td>
<td>SkyTrac ISAT IAirborne Position / Data Communicator Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
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<td>STC Holder</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Data for installation of configurations A to F, Aeron Design Ltd. Flight Manual Supplement FMS764.91, Revision 4, dated 23 July 2014, Aeron Design Ltd. Instructions for Continued Airworthiness ICA794.90, Revision 6, dated 15 July 2014. For configuration G - Quick Release Cargo Basket Installation Dart Aerospace Ltd.</td>
<td>Approved type of operation - VFR only. For AS 355, CAT A operations are forbidden. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/10/2021</td>
<td>Active</td>
<td>EASA 10240495</td>
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<td>TCCA SH08-16, ISSUE 7</td>
<td>Installation of External Attachment Provisions as detailed below. Configuration A: External Attachment Provisions Only - Installation of External Attachment Provisions to be completed in accordance with TCCA approved, Aeron Design Ltd. Document Control List DCL766-1, Revision 4, dated 17 July 2014, or later.</td>
<td>For configuration 1A - Aeron Design Ltd. Flight Manual Supplement FMS762.91, Revision 0, dated 30 June 2016; Aeron Design Ltd. Instructions for Continued Airworthiness ICA1002.90, Revision 0, dated 12 July 2016. For configuration 2A - Dart Aerospace Ltd. Installation Instructions IN-DD50-1081, Revision A, dated 10 June 2016.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/10/2021</td>
<td>Active</td>
<td>EASA 10240495</td>
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<td>TCCA SH16-29, ISSUE 03</td>
<td>Installation of Quick Release Bicycle Racks Installation of Quick Release Bicycle Racks to be completed in accordance with TCCA approved, Aeron Design Ltd. Document Control List DCL1002-1, Revision 0, dated 12 August 2016, or later approved revision.</td>
<td>For configuration 1A - Aeron Design Ltd. Flight Manual Supplement FMS762.91, Revision 0, dated 30 June 2016; Aeron Design Ltd. Instructions for Continued Airworthiness ICA1002.90, Revision 0, dated 12 July 2016.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/11/2021</td>
<td>Active</td>
<td>EASA 10240495</td>
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<td>TCCA SH09-38, ISSUE 6</td>
<td>Installation of Quick Release Maintenance Step Installation of Fixed Calm Step as detailed below. Configuration A: Quick Release Maintenance Step Installation of External Attachment Provisions in accordance with STC SH08-16. (Configuration A) is required for installation of External Attachment P...</td>
<td>For configuration 1A - Aeron Design Ltd. Flight Manual Supplement FMS762.91, Revision 0, dated 30 June 2016; Aeron Design Ltd. Instructions for Continued Airworthiness ICA1002.90, Revision 0, dated 12 July 2016.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/11/2021</td>
<td>Active</td>
<td>EASA 10240495</td>
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<tr>
<td>CAA UK RESPONSIBIL ITY</td>
<td>Introduction of Honeywell TCAS 7.1 to Boeing 737/738 design change entails the upgrade of the existing TCAS II processor Version 7.0 to 7.1</td>
<td>Leading Edge Aircraft Flight Manual Supplement, 2XL027 LEF71B (1) Issue 3 or later revisions of the above listed documents approved by EASA Leading Edge Master Data List, 2XL027 LEF781 (1) Issue 3.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/12/2016</td>
<td>Active</td>
<td>EASA 10260505</td>
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<tr>
<td>CAA UK RESPONSIBIL ITY</td>
<td>Temporary allocation of controller pilots and data link communications (CPDLC) for flight management.</td>
<td>Flight Manual Supplement S21 FMS-0114 Issue 3 or later versions of the above listed documents approved by EASA Master Document List S21 MDSL-1071 Issue 1.</td>
<td>None.</td>
<td>16/12/2016</td>
<td>Active</td>
<td>EASA 10260505</td>
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**List of EASA Supplemental Type Certificates**

easa.europa.eu

**Expiration:**

- **STC Holder:**
  - **DART AEROSPACE LTD.**
  - **TCCA AEROSPACE LTD.**
  - **2 EXCEL DESIGN LIMITED**
  - **STC TWENTY ONE LIMITED**

**Model:**

- **AS 355 F2, AS 355 N, AS 355 NP, AS 355 E, AS 355 F, AS 355 F1**

**Rev.:**

- **REV. 3**
- **REV. 4**
- **REV. 2**

**Date of Issue:**

- **16/12/2016**
- **26/10/2021**
- **10/11/2021**
- **16/12/2016**

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**Limitations/Conditions:**

- **Approved type of operation - VFR only. For AS 355, CAT A operations are forbidden. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.**
<table>
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<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>10060508</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 6</td>
<td></td>
<td>A340-341, A340-342, A340-343</td>
<td>STC Twenty One Ltd Aircraft Flight Manual Supplement 2S1 FMS-0143 Issue 1 Refer to STC Twenty One Ltd Aircraft Flight Manual Supplement Manual S21.44-14-1022, approved by EASA.</td>
<td>Certification basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Pre-condition of the installation of this design change is the modification of an aeroplane to incorporate either EASA STC 100497/14 (for A300/330 airframes) or EASA STC 100287/0 (for A340 airframes). Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair is determined.</td>
<td>11/05/2023</td>
<td>Active</td>
<td>10060503</td>
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<tr>
<td>10060520</td>
<td>INTELSAT INFLIGHT LLC</td>
<td>REV. 1</td>
<td>FAA STC ST04369RY</td>
<td>747-400</td>
<td>Installation of Gogo 2KU – System Activation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Configuration A and B of Design Change 2S1.44-14-1022, approved by EASA. The STC is to be used only with the tablets listed in chapter 0.6.2 of Instructions for Continued Airworthiness ICA-16-04. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair is determined.</td>
<td>28/01/2020</td>
<td>Active</td>
<td>10060520</td>
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<tr>
<td>10060524</td>
<td>ANUVU</td>
<td>REV. 2</td>
<td></td>
<td>EUR 190-300, EUR 190-200 STD, EUR 190-200 LR, EUR 190-200 IGW, EUR 190-100 LR, EUR 190-100 SR, EU 190-100-100 IGW, EUR 190-100 ECJ</td>
<td>- Master Document List No. MDL-16-06-14 Revision 4, dated 22 May 2019; - Master Document List No. MDL-1331 Issue 1 Refer to STC Twenty One Ltd Aircraft Flight Manual Supplement Manual S21.44-14-1022, approved by EASA.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>The STC is to be used only with the tablets listed in chapter 0.6.2 of Instructions for Continued Airworthiness ICA-16-04. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair is determined.</td>
<td>04/09/2020</td>
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<td>10060524</td>
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<td>10060527</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>REV. 1</td>
<td>FAA STC S021243E</td>
<td>SEE FAA AIM</td>
<td>Installation of Garmin transponders: GTX 345/35R with ADS-B Out and Informativity; GTX 35S/35R with ADS-B Out functionality. Revision 1: Add EC135 P3 and EC135 T3 models to the FAA Approved Model List.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Compliance of this installation with CS-ACNS has not been demonstrated. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aeroplane.</td>
<td>27/08/2020</td>
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<td>10060530</td>
<td>SPECIALIST AVIATION SERVICES LTD</td>
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<td>AW169</td>
<td>Flight Manual Supplement FMS981-043 Issue 8 dated 08/11/2016</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>This STC is applicable only to AW169 Serial Numbers complying with theavisce: configuration described in SAS documents MPS/881-040 Issue A and FTFRH/021 Issue A.</td>
<td>19/12/2016</td>
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<td>STC Holder</td>
<td>STC no.</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
<td>10060536</td>
<td>Rev I</td>
<td>206B</td>
<td>Leica ALS Installation on C208B equipped with EASA STC 10015027 Rev. 1 Transfer to ACC Columbia Jet Service GmbH</td>
<td>According to GFM Master Document list MD01L_2040415_rev0_20161020. To be updated in accordance with Gomolzig</td>
<td>No. SP DOH_2040419_rev0_20161020 or later revisions of the above listed documents approved by EASA. To be installed in BELL HELICOPTER 206A, 206B, 206L, 206L-1, 206L-3, 206L-4, 206, 206A, 206B, 206L, 206L-1, 206L-3, 206L-4.</td>
<td>31/12/2022</td>
<td>Active</td>
<td>10060536</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>10060542</td>
<td>FAA STC</td>
<td>206L-3, 206L-4, 206A, 206B, 206L-1</td>
<td>Installation of S-TEC System 20/30 Automatic Flight Control system.</td>
<td>According to FAA STC Revision 1. To be installed in accordance with FAA STC approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/12/2016</td>
<td>Active</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>10060550</td>
<td>FAA STC</td>
<td>206L-3, 206L-4, 206A, 206B, 206L-1</td>
<td>Garmin GNS 430W and GTX 330 installation.</td>
<td>- Bulletin No. 778, dated 14 May 1997.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/12/2016</td>
<td>Active</td>
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<td>AVIATION SUPPORT GmbH</td>
<td>10060571</td>
<td>FAA STC</td>
<td>206L-3, 206L-4, 206A, 206B, 206L-1</td>
<td>Installation of EMS Equipment SA 365N installation of EMS Equipment SA 365N series helicopters.</td>
<td>- Flight Manual Supplement, FMS- ASE-91135, original release and alternative or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/12/2016</td>
<td>Active</td>
<td>10060571</td>
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<td>HONEYWELL INTERNATIONAL INC.</td>
<td>10060572</td>
<td>FAA STC</td>
<td>206L-3, 206L-4, 206A, 206B, 206L-1</td>
<td>Installation of the Bendix/King KFC 225 Automatic Flight Control system.</td>
<td>For the GAB model Master Drawing List 159-08311-001, Revision E, dated 12-19-2006 or later FAA approved revision For the GAB-TC320 model Master Drawing List 159-08311-001, Revision G, dated 1-3-2012 or later FAA approved revision Airplane Flight Manual Supplement, Honeywell part number 006-10320-001.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. This approval is limited to GAB aircraft.</td>
<td>21/12/2016</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10060575</td>
<td>STC TWENTY ONE LIMITED</td>
<td>FAA STC</td>
<td>RRU-95B</td>
<td>Introduction of a VIP Interior on Sukhoi RRU-95B.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s) (D-12 for leased compartments/Equivalent Safety Findings) F-34 for Quantity of Oxygen Determination/Deviation (s) (D-13 Internal Doors) The requirements f...</td>
<td>- Flight Manual Supplement ST21.FMS-019, Issue 1 - Master Document List ST21.MDL-1266, Issue 2 - Introduction of VIP Interior - Supplement Manual ST21.TEC-0474, Issue 1</td>
<td>This modification is applicable only to the RRU Serial Number 55060.</td>
<td>21/12/2016</td>
<td>Active</td>
<td>10060575</td>
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<td>10060577</td>
<td>AIRCRAFT SYSTEMS AND MANUFACTURING G, INC</td>
<td>FAA STC</td>
<td>ST04027CH</td>
<td>Certification of the aircraft data link system used for communications supporting Air Traffic Services (ATS) and installation of the Rockwell Collins Communications Management Unit (CMU), with a complementary Iridium Salcom system, or with the integration of an existing Salcom systemRev. 1 updates the de...</td>
<td>The Certification Basis for the original product remains applicable to this certification approval, except where amended by additional or later amendments as noted. The requirements for environmental protection and the airworthiness identified noise and/or emissions levels of the or...</td>
<td>1. Aircraft Systems (s) and/or the original product remains applicable to this certification approval.</td>
<td>05/08/2018</td>
<td>Active</td>
<td>10060577</td>
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<td>10060578</td>
<td>SPAES GmbH &amp; Co. KG</td>
<td>Rev. 3</td>
<td>AS 350 B2 A3 350 B3</td>
<td>AS 350 B2 A3 350 B3</td>
<td>Max Plot View Kit consisting of a new RH bubble door with a larger/breathable window which the disconforms to, a floor viewing aperture outboard of the RH pilot’s seat and a corresponding transparent section in the underlying fairing Rev. 1 allows some specific confi...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated identified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>Flight Manual Supplement, doc. Ref. GVH-6767-01-FMS-01 issue 5, dated 07/2017 by the FAA and/or later revisions of the above listed documents approved by EASA - Certification Programme GVH-6767-01-CP-01 issue 4, dated 20 December 2016 (for initial Issue) - Certification Programme GVH-6767-01-CP-02 issue</td>
<td>12/10/2022</td>
<td>Active</td>
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<td>10060586</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A319-111</td>
<td>Hyperion Cabin Modification</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition(s) CS25.301 Amdt. 1, CS25.303 Amdt. 1, CS25.305 Amdt. 1, CS25.307 Amdt. 1, AR25.365 Ch14/Ch13, CS25.601 Amdt. 0, CS25.603 Amdt. 10, CS25.605 Amdt. 0, CS25.609 Amdt. 0, CS25.611 Amdt. 0, The requirements for environmental protection and the associated identified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>- Supplemental Type Definition Document, Ref. STD225-1637M Issue 1 Rev.0 - Instructions for Continuous Airworthiness, Ref. A-03-1676-7A-CA-14 Issue 1Rev. O-Compliance Record List, Ref. CR532-1637M Issue 1 Rev. O-3 for later revisions of the above listed documents approved by EASA</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will not introduce roadways effect upon the overall product.</td>
<td>22/12/2016</td>
<td>Active</td>
<td>10060586</td>
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<td>10060587</td>
<td>BOMBARDIER Inc.</td>
<td>Rev. 1</td>
<td>TCCA STC G-LSA99-055G ISSUE 2</td>
<td>CL-600-2B16 (601-3A VARIANTS)</td>
<td>Installation of Oxygen System Pressure Discharge Indicator</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated identified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>- TCCA Supplemental Type Certificate (STC) No. G-LSA99-055G Issue No. 2, dated 11th November 2016 - Ksix Aviation Inc. Master Document List, Ref. doc. MDS5118004, Revision A, dated 15th November 2016 - Ksix Aviation Inc. Compliance Plan Document, Ref. doc. CP36118-001, Revision A, dated 17th...</td>
<td>Applicable to CL-600-2B16 (601-3A Variant), Aircraft SeralNo. 5125</td>
<td>22/12/2016</td>
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<td>EASA STC no</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC no</td>
<td>Model</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10080588</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>10058</td>
<td>10060590</td>
<td>JETSTREAM 3200 SERIES</td>
<td>GPS Antenna Installation – Two antenna on upper fuselage (STA 133 onboard BL 7L and BL 3R)</td>
<td>The Certification Basis for the original product as amended by the installation of the new equipment must be maintained in accordance with Document ID 2104016 STC, Issue 1, Dated 02/12/2016</td>
<td>As specified in document ID 2104016-CA Issue 1</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>22/12/2016</td>
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<td>10080590</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10059</td>
<td>10060624</td>
<td>GIV</td>
<td>Installation of Honeywell Primus Elite Advanced Features (PEAF) Upgraded DU-885 Displays.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental production and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Gulfstream Aerospace Corporation Index List C/E412088001, Rev. A, dated March 30, 2016, or later FAA approved revision 1. For-401 Configuration-FAA accepted Instructions for Continued Airworthiness CE412088A001, Revision B, dated October 23, 2014, or later FAA approved revision subsequent to Document ID 204016-CA Issue 1.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>22/12/2016</td>
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<td>10080591</td>
<td>AEARO TECHNOLOGIES LLC</td>
<td>10059</td>
<td>10060622</td>
<td>A340-300 SERIES</td>
<td>Installation of thermal acoustic insulation packages</td>
<td>The Certification Basis (CB) for the original product as amended by the installation of the new equipment must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>Aearo Technologies Master Data List MDL-1627-033015-001, Revision A, dated 20 July 2018</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>22/12/2016</td>
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<td>10080622</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>10062</td>
<td>10060624</td>
<td>MBB-BK117 D-2</td>
<td>Installation of External Life Raft System (ELRS)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental production and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>MBB-BK117 C-2 Master Document List (MDL) MDL-145-3, Revision G, dated 17 November 2016, or later FAA approved revision Instructions for Continued Airworthiness (CA) ICA/145-3, Revision A, dated 16 November 2016, or later FAA accepted revision.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>03/01/2017</td>
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<td>10080624</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>10062</td>
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<td>03/01/2017</td>
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<th>Foreign/NAA Code</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0426WAT-D</td>
<td>G150</td>
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<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-AO/51.01</td>
<td>Airplane Flight Manual Supplement Document No. G150-CA217-005. Rev. B, dated 21 October 2019. Index List Doc No. CEC110683, Rev. B, dated 13 May 2019. Instructions for Continued Airworthiness Doc. No. G150-CA217-004, Rev A, dated 16 December 2019. The latest revision of the above listed documents.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise/misuse on the aircraft or its product.</td>
<td>03/07/2023</td>
<td>Active</td>
<td>10060626</td>
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<tr>
<td>BHE &amp; AIRBUS</td>
<td>FAA SA113626C</td>
<td>B200, B200C, B300, B300C</td>
<td></td>
<td>Installation of a Rockwell Collins Pro Line Fusion II embedded displaysystem (EDS) and optional TCAS III. Rev. 3 updates the existing Collins Aerospace Pro Line 21 system on B300/B300C.</td>
<td>Refer to FAA Approved Model List (AML) STC Number SA113626C. Rev. F dated 14 February 2023. Software update in accordance with BHE &amp; ASSOCIATES, LTD Service Bulletin (SB) of 522-05-0142 Rev IR, dated 16 August 2019, for B300 and B300C.. Software update in accordance with BHE &amp; ASSOCIATES, LTD Service Bulletin.</td>
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<td>11/05/2023</td>
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<td>S-TEC CORPORATION</td>
<td>FAA STC SA8945W-D</td>
<td>F172M, F172N, F172P</td>
<td></td>
<td>Installation of a S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-55X.</td>
<td>Refer to FAA Approved Model List (AML) STC Number SA11136SC Rev. F dated 14 February 2023 for applicable serial numbers. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will not introduce noise/misuse on the aircraft or its product.</td>
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<td>04/01/2023</td>
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<td>AIRBUS HELICOPTERS DEUTSCHLAND GmbH</td>
<td>FAA STC SA113626C</td>
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<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-AO/51.01</td>
<td>Flight Manual Supplement GTX335R Installation SK-MAC-167-011R.</td>
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<td>09/10/2023</td>
<td>Active</td>
<td>10060634</td>
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<td>EMBRAER ENGINEERING TECHNOLOGY &amp; NTER USA, INC.</td>
<td>FAA STC SA113626C</td>
<td></td>
<td></td>
<td>Installation of a Garmin Transponder GTX335R with special Mission Configuration for Mode S and ADS-B in accordance with compliance report SK-MAC-167-011R.</td>
<td>The helicopter must be already equipped with a suitable GPS/Deziation module. A. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce noise/misuse on the aircraft or its product.</td>
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<td>25/11/2019</td>
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<td>Model</td>
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<td>Limitations/Conditions</td>
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<td>T303</td>
<td>MTV-14 Propeller on Cessna T303.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-36, Amendment 3; FAR 36, Amendment 36-30; This certificate/approval involves a change to the requirements environment.</td>
<td>1. STC Data List, ASIG Document No. 1065-3005-00, Rev. IR, dated 5/3/2016, or later FAA approved revisions.2. Installation Requirements Assessment, ASIG Document No. 1065-3005-00, Rev. IR, dated 2/24/2016, or later FAA approved revisions.</td>
<td>1. The SkyTender Beverage Trolley must be installed in accordance with installation Requirements Assessment (&lt;&lt;&gt;&gt; ASIG Document No. 1065-3005-00, Rev. IR, dated 2/24/2016, or later FAA approved revisions.2. Maintenance Instruction 1065-208009, dated February 25 2016, rev. A is a required part of this installation.</td>
<td>05/01/2017</td>
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<td>10260460</td>
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<td>A319-132</td>
<td>Cabin Retrofit by Seat Replacement and Soft Cabin Items Refurbishment.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Initial The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-36, Amendment 3; FAR 36, Amendment 36-30; This certificate/approval involves a change to the requirements environment.</td>
<td>None</td>
<td>06/01/2017</td>
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<td>10000060</td>
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<tr>
<td>EC35 T1, EC35 T2+, EC35 T3</td>
<td>Data Transmission System DTSV2/E135 Revision 01- Installation of DTSV2 in accordance with STC Data List SDL-DTSV2-EC135-T3H</td>
<td>Data Transmission System DTSV2/E135 Revision 01- Installation of DTSV2 in accordance with STC Data List SDL-DTSV2-EC135-T3H</td>
<td>- STC Data List SDL-DTSV2-EC135-T3H</td>
<td>Limited to helicopters with CDS/CPDS cockpit variants. For EC35 T3 model also applicable to helicopters with Helionix cockpit variants. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change(s) and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/07/2023</td>
<td>Active</td>
<td>10260660</td>
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<td>A330-243</td>
<td>Installation of KA-Band SATCOM System.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-36, Amendment 3; FAR 36, Amendment 36-30; This certificate/approval involves a change to the requirements environment.</td>
<td>Supplemental Type Definition Document, Ref. STDO25-16/GM, Issue 1Rev. 0 - Compliance Requirements List, Ref. CRLL23-16/GM, Issue 1 Rev. 0 - Instructions for Continued Airworthiness, Ref. A-00-1675-ICA Issue 1Rev. 0 - Airplane Flight Manual Supplement, Ref. A-00-1675-4FM Issue 1 Rev. for later revisions...</td>
<td>Refer to the Instructions for Continued Airworthiness, Ref. A-00-1675-ICA Issue 1Rev. 0 (Section 4 Airworthiness Limitation Section) Prior to installation of this design change must be determined that the interrelationship between this design change and any other previously installed design change(s) and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/01/2017</td>
<td>Active</td>
<td>10260667</td>
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### List of EASA Supplemental Type Certificates

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<th>STC Holders</th>
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<th>FAA STC STC No.</th>
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<tr>
<td>McCauley Propeller Systems</td>
<td>1006070</td>
<td>FAA STC ST0674</td>
<td>PA-28R-201T</td>
<td>Installation of a McCauley Propeller Systems model 4HRFGCT78 hub and 1028HA-0 blade for a Textron Aviation Inc. model 208B aircraft with a P+V+/-CWC PT6A-140 turboprop engine.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval: NEW-PAGE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/01/2017</td>
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<td>1006070</td>
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<tr>
<td>MAG AEROSPACE INDUSTRIES, Inc./BA MONOGRAM SYSTEMS</td>
<td>1006074</td>
<td>FAA STC ST0288LA</td>
<td>MAG</td>
<td>Installation of Toilet Assembly.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/01/2017</td>
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<td>1006074</td>
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<td>Friionitec Ltd.</td>
<td>1006080</td>
<td>PA-28R-201T</td>
<td>AVIONTEC</td>
<td>Autonics RotorRemoval - Bendix King HMU-20 Auto Panel SPA 400 Intercom - Bendix King KR 85 ADF</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Limited to S/N 28R7703135 to prior installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/01/2017</td>
<td>Active</td>
<td>1006080</td>
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<tr>
<td>SAA Solutions For Aviation, S.L</td>
<td>1006086</td>
<td>FAA STC ST0674</td>
<td>S4A</td>
<td>Structural Modifications in accordance with S4A Master Drawing List 028834-RPT02, issue 1</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(h) at a later amendment: CS 23, Ampted A - 23.1,23.3, 23.21, 23.26, 23.261, 23.301, 23.303, 23.305, 23.307, 23.321, 23.337, 23.341, 23.471, 23.581, 2...</td>
<td>This STC is limited to the modifications as defined in Master Drawing List 028834-RPT02, issue 1, but does not cover any installation of additional equipment... Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously...</td>
<td>11/01/2017</td>
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<td>1006086</td>
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<tr>
<td>Cessna Düsseldorf Citation Service GmbH</td>
<td>1006088</td>
<td>FAA STC ST0674</td>
<td>CESSNA</td>
<td>Installation of Garmin GTX 725W.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- According to the section of AFM-S-0300 equipped S/N excluded to prior installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/05/2019</td>
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<td>STC Holder</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:CS-36, Amendment 3; FAR 36, Amendment 36-307; this certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels of the original product.</td>
<td>According to MT-Propeller Master Documentation List No. E-2833, Rev 0; To be operated in accordance with MT-Propeller AFM-S Document No E-2833, Rev 0 or later revisions of the above listed documents approved by EASA. To be installed i.e. Installation Instruction Doc. No. E-2834, Rev 0 or later revisions.</td>
<td>Acceding limitation section of AFM-S. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/01/2017</td>
<td>Active</td>
<td>10060601</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Supplemental Type Definition Document, Ref. STD023-16/2M Issue 1, Rev. 1; Compliance Record List, Ref. CRLL23-16/2M, Issue 1 Rev. 0; Airplane Flight Manual Supplement, Ref. A-00-16/61-AP, Issue 1 Rev. 2; Instructions for Continued Airworthiness, Ref. A-00-16/81-CA, Issue 1 Rev 0 or later revisions...</td>
<td>Requires the previous installation of the following modifications: Airbus Service Bulletin A330-23-3391 and A330-23-3267. EASA STCs 10028128 and 10060681. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/01/2017</td>
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<td>10060693</td>
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<td>BOMBARDIER Inc.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Modification to be completed in accordance with BE Aerospace Inc Modification Data Summary number L179000 Revision NC&lt;.&lt;.&lt;&gt;.&lt; or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EUL Canada/Bilateral Agreement.</td>
<td>None</td>
<td>13/01/2017</td>
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<td>10060703</td>
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<td>ALPINE AEROTECH LP</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Instructions for Continuing Airworthiness Document Number A-01-015-701 Revision A, dated 20 November 2015. Installation Instructions Document Number AAL-350-015-001 Revision N dated 01 June 2016. Master Data List Document Number AAL-350-014-601 Revision B, dated 01 June 2016.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/01/2017</td>
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<td>DART AEROSPACE Ltd</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Dart Instructions for Continued Airworthiness ICA-D139-1019, Rev. 0 or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EUL Canada/Bilateral Agreement. TCCA Supporting after C-16-0737, dated 5 August 2016. Dart Airworthiness Cam...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/01/2017</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10060712 DART AEROSPACE Ltd.</td>
<td>0</td>
<td>TCAS SH12-59</td>
<td>407 206L 206L-1 206L-3 206L-4 206B</td>
<td>Configuration 2Cyckle Stick Control Friction in accordance with Aero Design Ltd.DCI.953-1, Revision 2, dated 23 November 2016 or later TCAS approved revision. Configuration 2Cyckle Stick Control Friction in accordance with Dart Aerospace Ltd.DML.D407-1086, Revision A, dated 17 May 2020 or later TCACA, ep...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Configuration FAA Design Flight Manual Supplement FM5092-1, Revision 1, dated 09 August 2014. FAA Design Instruction Number Continued Airworthiness ICA 952.90, Revision 1, dated 09 January 2015. Configuration 20AR Aerospace Ltd Installation Instructions INI-C407-1086 Revision A, dated 31 March 2020. ...</td>
<td>Prior to installation of the design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/ noise effect upon the airworthiness of the product.</td>
<td>10/11/2021</td>
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<td>10060718 GULFSTREAM AEROSPACE Corp.</td>
<td>0</td>
<td>FAA ST04207AT-D</td>
<td>G280</td>
<td>Installation of an Aircraft High Speed Internet System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Aircraft Flight Manual Supplement GA14320828-1, Revision Dated 24 October 2012. - Gulfstream Aerospace Corporation Master Data List GA14320820-1, Revision P, Dated 12 of November 2012. - Instructions for Continued Airworthiness GA143203A40, Revision C, Dated 26 October 2012 or later revisions of the above listed documents approved by EASA.</td>
<td>The following STC is a required part of STC 10052498. Installation of an Executive Cabin Interior. This system is intended to provide Internet connection and entertainment services using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination of the certificate/approval.</td>
<td>16/01/2017</td>
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<td>10060727 JET AVIATION AG</td>
<td>0</td>
<td>FAA ST03802-20A</td>
<td>A330-200, A330-300</td>
<td>Quad. Zoom and Tail Camera Installation Revision 1: Transfer of EASA STC 10060727</td>
<td>See appendix J:SC- M-0088CCS-01, Issue 03. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Compliance Sheet: J:SC- M-0088CCS-01, Issue03. Master Document List: J:SC- M-0088MDL-01, Issue 02. - Instructions for Continued Airworthiness: J:SC- M-0088ICA-01, Issue01. J:SC-M-0088ICA-02, Issue 01 for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/ noise effect upon the airworthiness of the product.</td>
<td>22/01/2017</td>
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<td>10060730 EDGE AERODYNAMIX, INC</td>
<td>0</td>
<td>FAA STC ST05923LA</td>
<td>737-800 900ER 737-300 400/500-600-700-7</td>
<td>Installation of Conformal Vortex Generators.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA approved Edge Aerodynamix, Inc. Master Drawing List E51001-1, Revision D, Dated 26 October 2016 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/ noise effect upon the airworthiness of the product.</td>
<td>24/01/2017</td>
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<td>10060733 ALPINE AEROTECH LP</td>
<td>0</td>
<td>TCAS SH12-36</td>
<td>2.205A 1 2.121 2.121 412EP</td>
<td>Installation of Baggage Compartment Protection Kit defined with Alpine Aerotech Master Data List AAL-221-054-001.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Alpine Aerotech Master Data List Document Number AAL-221-054-001.Revision D, Dated 26 August 2016. - Alpine Aerotech Installation Instructions Document Number AAL-221-055-005 Revision N, Dated 17 July 2016. - Alpine Aerotech Instructions for Continuing Airworthiness Document Number AAL-221-055-701 R.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/ noise effect upon the airworthiness of the product.</td>
<td>18/01/2017</td>
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<td>10060742 BELL HELICOPTER TEXTRON Inc.</td>
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<td>FAA STC SR00044AT</td>
<td>206L-4 206L-3 206L-1 206L 206B 206A</td>
<td>Installation of replacement battery door</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Bell Aircraft Accessories, Inc. Drawing List Report No. A-92154, revision, dated November 16, 1992, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/ noise effect upon the airworthiness of the product.</td>
<td>20/01/2017</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
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<tr>
<td>10060769</td>
<td>KORFF LUFTFAHRT GMBH</td>
<td>Increase of MTOW up to 850 kg (including new increased wingspan), nosepropeller, various improvements reg. instruments.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>Not possible for the SN: 6001 and 6010 of the model Grob G103. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise or emissions effects upon the airworthiness...</td>
<td>20/12/2017</td>
<td>Active</td>
<td>10060769</td>
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<td>10060774</td>
<td>AIRCRAFT SPECIALTIES SERVICES, Inc.</td>
<td>Regrind Teledyne Continental crankshafts, to 0.02” undersize on crankpin/id roll journals</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Specialties Services, Inc. Drawing No. 1010, Rev. C, dated 01/December 2004, or later FAA approved revision.</td>
<td>23/01/2017</td>
<td>Active</td>
<td>10060774</td>
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<tr>
<td>10060790</td>
<td>STANDARD AERO BUSINESS AVIATION SERVICES, I.LLC.</td>
<td>Installation of a Honeywell JetView System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Standard/Aero Master Data List 1033667, Revision (E), dated 19/November 2016 or later FAA approved revision. 2. FAA Approved Airplane Flight Manual Supplement document number 1033663, Rev (B) dated 8 November 2016, or later FAA approved revision 3. Standard/Aero Instructions for Continued Airworthiness...</td>
<td>03/02/2017</td>
<td>Active</td>
<td>10060790</td>
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<tr>
<td>10060791</td>
<td>AVIATION FABRICATORS, Inc.</td>
<td>Installation of a stretcher assembly onto a cabin seat track</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Aviation Fabricators STC Drawing List, Document No. AF-596, Rev B, dated 29 April 2016 or later FAA approved revision 2. Aviation Fabricators Airplane Flight Manual Supplement (AFMS) Document No. AF-596AFMS, Rev: IR, dated 19 October 2016, or later FAA approved revision.</td>
<td>25/01/2017</td>
<td>Active</td>
<td>10060791</td>
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<tr>
<td>10060794</td>
<td>JET AVIATION AG REV. 1</td>
<td>See appendix, JBC - M-0051CCS-01 Issue 04-The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Aircraft Flight Manual Supplement: JBC - M-0051APM-01 ISSUE 4- Master Document List: JBC-M-0051MDL-01 Issue 3 - Compliance Check List: JBC - M-0051CCS-01 Issue 4- Instructions for Continued Airworthiness: JBC - M-0051CA-01 Issue 3 or later revisions of the above listed documents approved and...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise effect upon the airworthiness of the product...</td>
<td>06/10/2017</td>
<td>Active</td>
<td>10060794</td>
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The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

- Supple\ntimental Type Definition Document, Ref. STC026-15/36/Issue 1, Rev. 0- Instructions for Continued Airworthiness, Ref. A-00-16/89/ICA Issue 1, Rev. 0, Compliance Record List, Ref. CRL25-15/36/Issue 1, Rev. For later revisions of the above listed documents approved by EASA.

- This STC requires the pre-installation of the delivery layout DLM 350-25-46/11 (certified by Airbus with EASA Major Change with Approval Ref. 10006442) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously insta...
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<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<tr>
<td>10060824</td>
<td>JET AVIATION AG, BASEL COMPLETIONS</td>
<td>REV. 3</td>
<td>A330-243</td>
<td>VIP Cabin Interior Compartments Rev. 2 Transfer of STC Rev. 3: Activation of flight monitors in cabin.</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>STC ST.7500 Crash Resistant Fuel Certification Basis</td>
<td>FAA STC M-00056AFM-01, Issue 01; M-00056MLD-01, Issue 05; CCS: JBSC-M-00056CCS-01, Issue 04; ICA: JBSC-M-00056ICA-01, Issue 01; JBSC-M-00056ICA-02, Issue 01; JBSC-M-00056ICA-03, Issue 02; JBSC-M-00056ICA-04, Issue 01;</td>
<td>06/11/2017</td>
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<tr>
<td>10060826</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 1</td>
<td>690D, 695, 695A, 695B</td>
<td>ASTM-27</td>
<td>On Twin Commander 690, 695 Series</td>
<td>The Certification Basis for the original product as amended, the following additional or alternative environmental requirements: the following paragraph(s) at a later amendment: CS-96, Amendment 4FAR 36, Amendment 36-30: This certification approval involves a change to the requirements for environmental...</td>
<td>According to MT-Propeller Master Documentation List No. E-2962. Rev 0 To be operated in accordance with MT-Propeller FAA-S Document No. E-2955, Rev 0; later revisions of the above listed documents approved by EASA To be installed i.a.w. Installation Instruction Doc. No. E-2960, Rev 0 or later Revs.</td>
<td>18/07/2017</td>
<td>Active</td>
<td>10060836</td>
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<td>10060831</td>
<td>PRECISE FLIGHT, Inc.</td>
<td>REV. 4</td>
<td>FAA STC SH3319NM</td>
<td>MBB-BK117 B-2C-1C-2/ B-2/ D-2</td>
<td>Installation of Precise Flight PulseWise Control Unit in the landing/take-off Recognition Visual Control Light System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. NEW-PAGE-The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Precise Flight Instructions as listed on FAA AML No. SH3319NM, dated 23 January 1986; or later FAA approved revision.</td>
<td>10/02/2017</td>
<td>Active</td>
<td>10060831</td>
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<tr>
<td>10060846</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>REV. 4</td>
<td>FAA AML SA01818WI</td>
<td>FAA AML SA01818WI</td>
<td>Installation of Garmin G5 Electronic Flight Instrument Rev. 1 amends the certification basis Rev. 2 updates the type data and adds more models to the AML Rev. 3 updates the type data and adds more models to the AML Rev. 4 amends the certification basis</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>FAA Approved revision.</td>
<td>The location of the pulselite box has not been evaluated; except if installation drawing manual FAA STC SA01818WI, dated 22 July 2016; or later FAA approved revision.</td>
<td>24/05/2023</td>
<td>Active</td>
<td>10060846</td>
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<td>10060852</td>
<td>AIRBUS HELICOPTERS</td>
<td>REV. 1</td>
<td>AS 350 B3</td>
<td>STC ST.7500 Crash Resistant Fuel System: Revision 01. Extension to EC130 B4</td>
<td>The Certification Basis for the original product as amended, the following additional or alternative environmental requirements: the following paragraph(s) at a later amendment: CS-27 Amdt. 3, dated 11 December 2012, 27.561(h), 27.636, 27.918(d), 27.952, 27.963(f), 27.967, 27.973, 27.975(b) for the...</td>
<td>Refer to 350NO42807E_TN “EC130/7500 Master Document List”, issue A, dated 12 December 2019 for pre-requisite installation/repair incompatibility list. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change...</td>
<td>27/01/2020</td>
<td>Active</td>
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<td>Status</td>
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<td>10060857</td>
<td>DAO AVIATION AVS</td>
<td>16/01/2018</td>
<td>10060881</td>
<td>None</td>
<td>Dual Garmin GTN 650 Installation on the JETSTREAM 3100/3200 SERIES.</td>
<td>C/S 23 Amendment 4. The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate approval.</td>
<td>Active</td>
<td>16/01/2018</td>
<td>10060857</td>
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<td>10060862</td>
<td>AERODATA AG</td>
<td>03/07/2024</td>
<td>10060881</td>
<td>None</td>
<td>Installation of a Garmin transponders (a) GTX 330/330D/33 series or GTX 335/335R/335D/335DR with ADS-B Out functionality (b) GTX 345/345R/345D/345DR with ADS-B Out and In functionality (c) GTX 355R/355DR with ADS-B Out functionality in select airplane installed with G950/G1000 systems; or (d) GTX 345R/345D.</td>
<td>Active</td>
<td>06/02/2017</td>
<td>10060862</td>
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<td>10060873</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>01/02/2017</td>
<td>10060881</td>
<td>None</td>
<td>Garmin Master Drawing List (MDL) 005-00734/04, Revision 1, dated 1 May 2013, or later FAA approved revision.</td>
<td>Installation and Operation according to the Master Document number MDL-AC-01B.</td>
<td>Active</td>
<td>02/10/2019</td>
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<td>10060874</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>02/10/2019</td>
<td>10060881</td>
<td>None</td>
<td>Garmin Master Drawing List (MDL) 005-00734/04, Revision 1, dated 1 May 2013, or later FAA approved revision.</td>
<td>Garmin Master Drawing List (MDL) 005-00734/04, Revision 1, dated 1 May 2013, or later FAA approved revision.</td>
<td>Active</td>
<td>20/12/2022</td>
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<td>10060881</td>
<td>E A D AEROSPACE AIRWORKINESS</td>
<td>01/02/2017</td>
<td>10060881</td>
<td>None</td>
<td>The Certification Basis for the installation of this change/repair is the FAA approved ADS-B position source. Refer to design data specified in the Master Drawing List (MDL) listed above for specific hardware and software requirements.</td>
<td>Aircraft installation involving the Garmin transponder model without an internal GPS require the previous installation of an approved ADS-B position source.</td>
<td>Active</td>
<td>01/02/2017</td>
<td>10060881</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<td>AIRBUS HELICOPTERS DEUTSCHLAND GMBH</td>
<td>Installation of Garmin G500H Electronic Flight Instrument system (EFIS), consisting of a primary and a multifunctional display, an altitude and heading reference system (AHRS), an air data computer (ADC), an artificial horizon and an outside air temperature (OAT) probe.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Flight Manual Supplement, doc. ref. SK-MAC-163- SJP-J-992 Revision C code 11-160 later revisions of the above listed documents approved by EASA. Certification Programme, doc. ref. SK-MAC-163-CPR-J-995 Revision C dated 22/06/2016. Compliance Report, doc. ref. SK-MAC-163-CRP-J-994 Revision D dated 22/06/2016.</td>
<td>- Garmin GTN650 equipment shall be installed as a prerequisite for this modification. - A standby attitude indicator must be installed. No new swords of operation are approved: Neither the Synthetic Vision System (SVS) nor the Helicopter Terrain Awareness and Warning System (HTAWS) functions have been c...</td>
<td>02/02/2017</td>
<td>Active</td>
<td>10000263</td>
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<tr>
<td>AERODATA AG</td>
<td>Enhancement of TDR94/94D transponders for ADS-B functionality. Upgrade of TDR94/94D transponders and connection of the transponders to both GPS receivers for compliance with RTCA DO-260B requirements for ADSB-out functionality.</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment: CS-25.404.4 and 4.4.9 of the following paragraph: 25.1309, 25.1317, 25.1529, 25.158A(k)(2) and (d), 25.158A(b)(1), (c)</td>
<td>Installation according to AERODATA Service Bulletin reference SB-H125-TDR-2158 Rev. 1&lt;(&gt;,&lt;) &gt;</td>
<td>- Operation according to AERODATA Aircraft Flight Manual Supplement reference AFM-158, Rev., Maintenance according to AERODATA Aircraft Maintenance Manual, Supplement, reference AAM-165, Rev. or later revisions.</td>
<td>27/06/2017</td>
<td>Active</td>
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<td>DART AEROSPACE Ltd.</td>
<td>Installation of Sidetrack Wearplate Kit P/N D350-1023.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>TCCA accepted Instructions for Continued Airworthiness ICA-D350-1023 Revd. dated 10th August 2016 (NOTE: no ALS changes have been introduced within STC) TCCA approved Installation Instructions (IN- D350)-1023 Rev A, dated 10th August 291671CCAP approved Master Document List MDD-D350-1023 Rev A, dated 1st Jan.</td>
<td>- Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2017</td>
<td>Active</td>
<td>10000264</td>
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<tr>
<td>JET AVIATION AG</td>
<td>Installation of Skidtube Wearplate Kit P/N D350-1023.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>TCCA accepted Instructions for Continued Airworthiness ICA-D350-1023 Revd. dated 10th August 2016 (NOTE: no ALS changes have been introduced within STC) TCCA approved Installation Instructions (IN- D350)-1023 Rev A, dated 10th August 291671CCAP approved Master Document List MDD-D350-1023 Rev A, dated 1st Jan.</td>
<td>- Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2017</td>
<td>Active</td>
<td>10000265</td>
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<tr>
<td>ANOU/AERONAUTIQUE S.R.L.</td>
<td>Indium Antenna Replacement. The modification installs a new Indium antenna replacing another one at the same location. The Revision 1 updates the limitation section, by removing the Indium design restriction.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List, ref : JB-IM-0086MLD-01 Issue 2, dated 31st September 2017. Certification for Continued Airworthiness ref : JBIL-M-0086ICA-1005 Issue 3, dated 31st September 2017.</td>
<td>- The installation of this modification is limited to MSN 5005-5 Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/01/2018</td>
<td>Active</td>
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<tr>
<td>ANOU/AERONAUTIQUE S.R.L.</td>
<td>A340-300/300/I3/CEB7/05 MODIFICATION The design change related to this STC consist in the replacement of existing safety belts by new ones manufactured by ANOU/AERONAUTIQUE Thermo fabricated on BC seats type 1404 &amp; EC seats type 9174 and 3509.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification Plan ref : 02-610850-CP-001 Rev 04- Master Data List ref : 02-610850-MLD-01 Rev 02- ICA- CMMS (Component Maintenance Manual Supplement), ref 02-610850-CMMS-003 Rev 1- Modification Approval Sheet, Project 02-610850 Rev 02(a) later revisions of the above listed documents approved by EASA.</td>
<td>- Certification Plan ref : 02-610850-CP-001 Rev 04- Master Data List ref : 02-610850-MLD-01 Rev 02- ICA- CMMS (Component Maintenance Manual Supplement), ref 02-610850-CMMS-003 Rev 1- Modification Approval Sheet, Project 02-610850 Rev 02(b) later revisions of the above listed documents approved by EASA.</td>
<td>07/02/2017</td>
<td>Active</td>
<td>10000267</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>10060941</td>
<td>AEROTEÇ &amp; CONCEPT</td>
<td></td>
<td>EC 225 LP</td>
<td></td>
<td>Starting System Adaptation with “Start Active” Flt Kit (PN EPA-AE-14121-100)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Robinson Flight Manual Supplement «Start Active» Flt Kit (PN EPA-AE-14121-100)</td>
<td>This design change cannot be installed on HIC with APU or with MOD404/4UMS’ sight. Prior to installation of this design change it must be determined that the interrelationships between this design change and any other pre-existing installed design change and/or repair will introduce no adverse effect upon...</td>
<td>08/02/2017</td>
<td>Active</td>
<td>10060941</td>
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<tr>
<td>10060944</td>
<td>KAMAN CORPORATIONALRCRAFT WHEEL AND BRAKE, LLC</td>
<td>REV. 1</td>
<td>FAA STC S436C3</td>
<td>B300C/ B300/ 300</td>
<td>Replace existing nose wheel with Cleveland Nose Wheel PN 45-204 by instaling Parker Hannifin Conversion Kit 199-126/Revision 1 is an administrative re-issuance following a name change from Parker Hannifin Corporation Aircraft Wheel and Brake Division to Kaman Corporation, Aircraft Wheel and Brake, LLC...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Parker Hannifin Conversion Kit 199-126, Revision M, dated 01 July 2011 or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU Bilateral Agreement. Prior to installation of this design change it must be determined that the interrelationships between this design change and any other pre-existing installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/06/2023</td>
<td>Active</td>
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<td>10060949</td>
<td>FLYING COLOURS CORPORATION</td>
<td>REV. 2</td>
<td>FAA STC 10297NY</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>AS350B3 Out Installation in the Bombardier CL-600-2B16 (CL-604 Variant)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- FAA STC 10297NY, dated 2nd October 2015- Dassault Aircraft Services Master Data List, ref.: doc. MDL000015, Revision C, dated 2nd February 2011 - Airplane Flight Manual Supplement, AFMS, ref. doc. FMS00012, Revision N, dated 2nd October 2015 - Instruction for Continued Airworthiness (ICA) ref. doc. This alteration requires as a prerequisite(&gt;),&lt; the Bombardier Inc. Service Bulletin 605-34-026 for Model CL-600-2B16, Installation of Global Positioning System (GPS) - Upgrade the Wide Area Augmentation System (WAAS) Capability to prevent signal loss during flight.</td>
<td>12/06/2017</td>
<td>Active</td>
<td>10060949</td>
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<tr>
<td>10060955</td>
<td>SABENA TECHNICS BOO</td>
<td></td>
<td>A340-313</td>
<td></td>
<td>IAGOS – RTTU installation This STC installs an additional module, named Real Time Transmission Unit (RTTU) to the integration of routine Aircraft measurements into aGlobal Observing System (IAGOS) on A340-313 aircraft model. This module receives IAGOS data and transmits them, through SATCOM system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Modification Approval Sheet, ref. S-23-2334, Issue 3 dated 07February 2017 - Instruction for Continued Airworthiness, ref. ICA-F043S-23-007 (issue 3), dated 22 October 2015 or later revisions of the above listed document(s) approved by EASA. 1- The modification is applicable to the following aircraft Serial Number(&gt;),&lt;...</td>
<td>09/02/2017</td>
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<td>10060955</td>
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<tr>
<td>10060962</td>
<td>SPAES GmbH &amp; Co. KG</td>
<td></td>
<td>EC135 T2/ EC135 T2+ EC135 P3/ EC135 T3/ EC135 T1 EC135 P1/ EC135 P2/ EC135 P0</td>
<td></td>
<td>Title: Installation of an automatic deployable ELT Description: For operation over water an automatic deployable ELT (AD-ELT) has to be installed. The beacon will be installed on the middlecover on the LH side of the lower fuselage shell. The cockpit controlpanel for activation and release will be in...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment: CS 27.25-1; 2. For models EC135 P1/(&lt;,&gt;)EC135 P2; EC135 T2; EC135 T2+; CS 27.21, CS 27.25, CS 27.27...</td>
<td>Supplement for Automatic Deployable ELT (AD-ELT), FMS- SPAES-0047 Revision 01 dated 02 February 2011 or later revisions of the above listed document(s) approved by EASA - Installation Manual, ICA-135-047-25-01, Ausgabe 08 dated 8 November 2016. Instructions for Continued Airworthiness, ICA-135-047-25-01, Ausgabe 08 dated 8 November 2016 - Airbus modifications 40907, 40982, 41082...</td>
<td>15/02/2017</td>
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<td>STC Holder</td>
<td>FAA STC/Document/Revision</td>
<td>FAA/AMS/Supplement</td>
<td>Serial/Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>LEARJET INC</td>
<td>FAA STC ST00370B</td>
<td></td>
<td>BD-700-1A11</td>
<td>Installation of a Honeywell Jetwave™ KA Band Satellite Data System.</td>
<td>1. Master Data List (MDL), Document number BAE87107033, Revision 0, dated 2 August 2016, or later FAA approved revision</td>
<td>1. Maintenance for the MDL-1 configuration must be in accordance with the FAA.</td>
<td>13/02/2017</td>
<td>Active</td>
<td>10060895</td>
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<td>HONEYWELL INTERNATIONAL (BENDIXKING)</td>
<td>FAA STC SA09658AC</td>
<td>SEE FAA AMS SA09658AC-2604/16</td>
<td></td>
<td>Installation of BendixKing Aerospace 100 In-flight Data Link System non-essential, non-required Aircraft Cabin Systems &amp; Equipments</td>
<td>1. Master Data List, Document No. 88800915-21, Revision 0, dated 16 March 2016, or later FAA approved revision</td>
<td>2. This STC is valid for aircraft types and model numbers:</td>
<td>02/03/2017</td>
<td>Active</td>
<td>10060879</td>
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<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>FAA STC AMS 3A45-100</td>
<td></td>
<td>A-340-1-100Property</td>
<td>Modification of Economy Class seats installed on VIP aircraft.</td>
<td>Certification Summary, Northwest Aerospace Technologies Inc, document No. N2500-500, Initial Issue, dated 2 February 2017</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and model numbers:</td>
<td>10/02/2017</td>
<td>Active</td>
<td>10060960</td>
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<td>CMD FLIGHT SOLUTIONS, LLC</td>
<td>FAA STC AMS 04515101</td>
<td>SEE FAA AMS</td>
<td>A-145515101</td>
<td>Certification of Automatic Dependent Surveillance - Broadcast (ADS-B) Out System with Rockwell Collins TDR-444/445 Transponders.</td>
<td>1. CMD Flight Solutions Master Data List, Document No. CB1500601, Revision JR, dated 19 June 2015, or later FAA approved revision. 2. FAA Approved Airplane Flight Manual Supplement (AFMS) as listed on the FAA AMS number SA04515101, or later FAA approved revision.</td>
<td>1. Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. 2. Additional Limitations and Conditions apply.</td>
<td>03/06/2017</td>
<td>Active</td>
<td>10061022</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
<th>FAA STC/ST04146CH</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>ARMSTRONG AEROSPACE Inc.</td>
<td>FAA STC ST04146CH</td>
<td>EASA validation of FAA STC ST04146CH. Installation of a Zodiac RAVE In-Flight Entertainment System in accordance with StandardAero Master Data List DC104-9752A-00, Revision (E) dated 02nd December 2016 or later approved revisions. NEW-PAGE</td>
<td>The Certification Basis for the original product as amended by subsequent or alternative airworthiness requirements CS 25 impacted requirements at initial issue unless otherwise specified 25.1a, b); 25.30/31a, i)(amendment 1); 25.303, 25.303(a)-(c) (amendment 1); 25.307(a) ) (amendment 1); 25.541a-(d) 2...</td>
<td>See those of original FAA STC ST04146CH and those of the Airplane FlightManual Supplement document number DC303-9752A-00, Revision (-) dated 19th October 2016, or later approved revisions. Prior to installation of this design change it must be determined that the interrelationship between this design change...</td>
<td>14/02/2017</td>
<td>Active</td>
<td>10061026</td>
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<tr>
<td>ANJOU AEROSPACE</td>
<td>FAA STC SA01672WI</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>The Certification Basis for the original product remains applicable to FAA STC SA01672WI. The requirements for environmental protection and the associated certified noise levels of the original STC installation must be performed in accordance with StandardAero Master Data List DC104-9752A-00, Revision (E) dated 02nd December 2016 and the Aircraft Flight Manual Supplement document number DC303-9752A-00, Revision (-) dated 19th October 2016 is a required part of this modification and the aircraft.</td>
<td>04/11/2022</td>
<td>Active</td>
<td>10061033</td>
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<tr>
<td>AVIATION FABRICATORS, Inc.</td>
<td>FAA STC SA01643SE</td>
<td>SEE APPROVED MODEL LIST</td>
<td>The Certification Basis for the original product remains applicable to FAA STC SA01643SE. The requirements for environmental protection and the associated certified noise levels of the original product remain applicable to this certificate/approval.</td>
<td>16/02/2017</td>
<td>Active</td>
<td>10061038</td>
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<tr>
<td>B.A.S., INC.</td>
<td>FAA STC SA01643SE</td>
<td>SEE APPROVED MODEL LIST</td>
<td>The Certification Basis for the original product remains applicable to FAA STC SA01643SE. The requirements for environmental protection and the associated certified noise levels of the original product remain applicable to this certificate/approval.</td>
<td>16/02/2017</td>
<td>Active</td>
<td>10061039</td>
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<tr>
<td>ANJOU AERONAUTIQUE S.R.L.</td>
<td>A330-243</td>
<td>Passenger Economic Seats Installation.</td>
<td>The Certification Basis for the original product as amended by subsequent or alternative airworthiness requirements for the following paragraph(s) at a later amendment: CS 25 Annex 16 for the affected area – refer to CP ref. 02-100009-CP-01 Rev. 0 for details. The requirements for environmental protection and the associated certified noise levels of the original product remain applicable to this certificate/approval.</td>
<td>16/02/2017</td>
<td>Active</td>
<td>10061050</td>
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<tr>
<td>AIRBUS HELICOPTERS</td>
<td>AS 350 B3, EC 130 B4</td>
<td>STC ST 7501 Rupture Resistance Fuel System. Revision 01. Extension to EC 130 B4</td>
<td>The Certification Basis for the original product as amended by subsequent or alternative airworthiness requirements for the following paragraph(s) at a later amendment: CS 25 Annex 16 for the affected area – refer to CP ref. 02-100009-CP-01 Rev. 0 for details. The requirements for environmental protection and the associated certified noise levels of the original product remain applicable to this certificate/approval.</td>
<td>24/01/2020</td>
<td>Active</td>
<td>10061056</td>
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<tr>
<td>CANNARD AEROSPACE CORPORATION</td>
<td>FAA STC ST02880CH</td>
<td>Installation of ATC Transponders with ADS-B Capability, LPV Approach Adapter, and FABS Upgrade.</td>
<td>The Certification Basis for the original product as amended by subsequent or alternative airworthiness requirements for the following paragraph(s) at a later amendment: CS 25 Amendment 4, 27 December 2007. 25.29(b); 25.30/31a, i)(amendment 1); 25.303, 25.303(a)-(c) (amendment 1); 25.307(a); 25.321a(x); 25.581(c); 25...</td>
<td>34-3475-06, Revision B, Airplane Flight Manual Supplement 4017f, dated 7th of November 2016, Document Control Drawing or later revisions of the above listed documents approved/accepted under the EASA system. Revisions of the above listed documents approved/accepted under the EASA system. Applicable to A330-243 MSN 0501 only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change or/and repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/02/2017</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
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<td>10061072</td>
<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td>AS 355 NP</td>
<td>0</td>
<td>AS 355 N</td>
<td>Installation of a video transmission system on AS355, mod number PP-082</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>PP-082-STC-MM2 Supplement to Aircraft Flight Manual, issue 2, dated 01Feb 2017 or later revisions of the above listed documents approved by EASA PP-082-ICA-00 Supplement to ICA (issue 0, dated 16 Feb 2017) No ALS update identified for this STC PP-082 MDL-03 &quot;Master Data list&quot;, issue 03, dated 16 Feb...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise illegally installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10061082</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>777-200</td>
<td>0</td>
<td>0</td>
<td>AZ – Cabin rearrangement: The modification consists of a cabin rearrangement with 293 passengers including SEVEN Economical Premium Economic Class seats installation and STELIA Business class seats installation.</td>
<td>The Certification Basis for the original product as amended by the following additional or/alternative airworthiness requirements: Special Condition (s) / CRI 0010044006 F-01, Non-rechargeable Lithium Battery Installation. The requirements for environmental protection and the associated/certified noise and/or emissions levels.</td>
<td>- Change Approval Sheet CAS-B777-16-007 rev 0-0 for Installation of Upper Class Suite with 293 passengers, including GEVEN Video Transmission Equipment qualification. It is required that:</td>
<td>Monitors and components of the IFE System, including In-Seat Power Supply System are treated as pure forms of mass underneath approval. Associated IFE sub-installations and Electrical connection to aircraft are not required further design approval including full equipment qualification. It is required that...</td>
</tr>
<tr>
<td>10061083</td>
<td>AIRBUS S.A.S.</td>
<td>A330-343</td>
<td>0</td>
<td>0</td>
<td>MOD CJ2041: &quot;MASTER MOD – VAA UCF: Installation of Upper Class Suite seats and and awning forward part of A330-343 cabin, with an additional bar unit. Activation of EFSistem.&quot;</td>
<td>The Certification Basis for the original product as amended by the following additional or/alternative airworthiness requirements: Special Condition (s) / CRI-D-3312-001 (Issue 2) Single Side Facing Seats. The requirements for environmental protection and the associated/certified noise and/or emissions levels.</td>
<td>- Certification Summary ref: ACJC-CAA-15-009 Issue 5-Maintenance Planning Document Supplement, ref: AE14266ICA Rev. 01, dated 17 February 2017. No ALS update identified for this STC PP-082 MDL-03 &quot;Master Data list&quot;, issue 03, dated 16 Feb...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise illegally installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10061089</td>
<td>AIR ALLIANCE GmbH</td>
<td>35'-35A/36-36A</td>
<td>0</td>
<td>0</td>
<td>TCA &amp; XDP Antenna Installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Instructions for Continued Airworthiness (ICA) doc. no. 2016_034_LJ35_007, revision 01, dated 17 February 2017. Master Document List (MDL), doc. no. 2016_034_LJ35_Form 2, revision 00, dated 20 February 2017 or later revisions of the above listed documents approved by EASA.</td>
<td>None.</td>
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<td>10061092</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>757-200/-300</td>
<td>0</td>
<td>0</td>
<td>Airborne wireless platform activation.</td>
<td>The Certification Basis for the original product as amended by the following additional or/alternative airworthiness requirements: for the following paragraph(s) at a later amendment CS25 as at: 15. 25.1301, 25.1309, 25.1353, 25.1431, 25.1529, 25.1581, 25.1588. The requirements for environmental protection.</td>
<td>Master Data List AE14266ML Rev. 0 dated 15/02/2017 Aircraft Flight Manual Supplement AE14266AFMS Rev. 0 dated 25/01/2017 Instructions for Continued Airworthiness AE14266ICA Rev. 1 dated 25/01/2017 or later revisions of the above listed documents approved by EASA.</td>
<td>As per AFMS - WiFi system is prohibited to work and must be switched OFF during takeoff, landing, and low visibility approach operation. Use &quot;WIFI OFF&quot; button located on CCP on the top part of A350 L/H Attendant’s area/air attendant seat. TPEIs (including GSM personal devices) are prohibited to work.</td>
</tr>
<tr>
<td>10061095</td>
<td>APICAL INDUSTRIES, Inc./d/b/a DART AEROSPACE</td>
<td>FAA STC SR0128LA</td>
<td>421-412EP</td>
<td>421-412EP</td>
<td>Installation of Apical Industries, Inc. Emergency Float Assemblies Partnumbers 20174, and 20715, in accordance with Master Document List MD412-1.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Instructions for Continued Airworthiness for the following additional or/alternative airworthiness requirements: for the following paragraph(s) at a later amendment CS25 as at: 15. 25.1301, 25.1309, 25.1353, 25.1431, 25.1529, 25.1581, 25.1588. The requirements for environmental protection.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise illegally installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>STC Holder</td>
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<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>ARCHEON HOLDINGS, L.L.C.</td>
<td>FAA STC ST04354AT</td>
<td>DHC-8-202 DHC-8-201</td>
<td>Installation of an Avionics, Inc. avWiFi wireless router system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for equipment/production and the associated certified noise and/ or emissions level of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Data List 150137-MDL, Revision C, dated February 2017 or later FAA approved revisions FAA approved Airplane Flight Manual Supplement for Archeion Holdings LLC, 150137-ARMS, Revision IR, dated 21 February 2017, or later FAA approved revisions</td>
<td>Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/ or repair will introduce a design change effect upon the airworthiness of the product.</td>
<td>28/02/2017</td>
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<td>AVIONIK STRAUSSING ENTWICKLUNGS GmbH</td>
<td></td>
<td>525</td>
<td>Installation of L-3 Communications Stormscope Antenna</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for equipment/production and the associated certified noise and/ or emissions level of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Documents List ASR-2016-006-MDL-01-00 dated 20/02/2017 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/ or repair will introduce a design change effect upon the airworthiness of the product.</td>
<td>24/02/2017</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST04300AT-D</td>
<td>GV-SP GIV-X</td>
<td>Installation of Satcom Direct Router (SDR) (This modification installs various configurations of single or duocommunication hub (SDR) together with several data systems (Ku system: Sat Broadband, GoGo Air To Ground) supporting voice and data communications.)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for equipment/production and the associated certified noise and/ or emissions level of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Index List, ref CE4142968001 Revision C, dated 12 May 2016 or later FAA approved Airplane Flight Manual Supplement forupea4142940500 Revision B, dated 09 May 2016: Aviation SCA-00930-ICA-R01 dtd. or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that there is an interrelationship between this design change and any other previously installed design change and/ or repair will introduce a design change effect upon the airworthiness of the product.</td>
<td>28/02/2017</td>
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<td>PMV ENGINEERING</td>
<td>REV. 1</td>
<td>ATR 72-212 A ATR 42-500</td>
<td>TPED tolerance on ATR42/72</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for equipment/production and the associated certified noise and/ or emissions level of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List SCA-00930-MDL-R01 dtd. 17 Feb 2017 Instructions for Continued Airworthiness SCA-00930-ICA-R01 dtd. 17 Feb 2017 or later revisions of the above listed documents approved by EASA.</td>
<td>Per Instructions for Continued Airworthiness SCA-00930-ICA-R01 dtd. 17 Feb 2017 or later revisions of the above listed documents approved by EASA.</td>
<td>06/03/2017</td>
<td>Active</td>
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<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>ARMSTRONG AEROSPACE Inc/dba ASTRONICS ARMSTRONG AEROSPACE</td>
<td>FAAC STC ST01678CH</td>
<td>777-200</td>
<td>EASA validation of FAA STC ST01678CH. Installation of a General Dynamics 10 VAC EmPower In-Seat Power Supply System in accordance with Armstrong Aerospace Master Data List. Doc. No. DC104-0491-00, Revision 10, dated 29th April 2016 or later approved revisions.</td>
<td>The Certification Basis for the original product as amended by applicable waivers.</td>
<td>The STC installation must be performed in accordance with Armstrong Aerospace Master Data List. Doc. No. DC104-0491-00, Revision 10, dated 29th April 2016, and The Aircraft Flight Manual Supplement Amendment 13.</td>
<td>See those of original FAA STC ST01678CH and of those of the Airplane Flight Manual Supplement Amendment 13.</td>
<td>28/02/2017</td>
<td>Active</td>
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<tr>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH16-17 ISSUE 1</td>
<td>A9139 AB139</td>
<td>D139-1012 Cargo Net Installations</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Dart Flight Manual Supplement FMS- D139-1012, Rev. Apr.</td>
<td>Prior to installation of this change it must be determined that the installation will not introduce a risk to the airworthiness of the product.</td>
<td>28/02/2017</td>
<td>Active</td>
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<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 2</td>
<td>767-300</td>
<td>Introduction of Digecor “Glide” IFE System B767Revision 1 removes the MSN Imulation Rev. 2. Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis for the original product as amended by applicable waivers.</td>
<td>AES Flight Manual Supplement No. 72 issue 1AES PRIMARY ICA DOCUMENT. - INTRODUCTION OF DIGECOR “GLIDE LITE” IFE SYSTEM BT76 No. AES-1009 Issue 1AES Master Data List AES-767DRLNDL Issue 11or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canadabiateral Agreement TCA. Supporting Letter C-16-0433, dated 2 May 2016.</td>
<td>Prior to installation of this change it must be determined that the installation will not introduce a risk to the airworthiness of the product.</td>
<td>28/04/2021</td>
<td>Active</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST03223CH</td>
<td>767-200, 767-300, 767-400ER</td>
<td>Certification of Compatibility with Cellular and/or Wireless Data Technologies.</td>
<td>The Certification Basis for the original product as amended by applicable waivers.</td>
<td>- Supplemental Type Certificate Project Results, Ref. GOA-2417-01, Rev. B, dated Jan. 13, 2017. Master Data List, Ref. DC104-9230-00 Rev. L-Aircraft Equipment Compatibility List, Ref. DC370-9230-01 Rev. G-Aircraft Equipment Compatibility List, Ref. DC370-9230-02 Rev. L-EWIS and ICA Assessment, Ref. DC01678CH and those of the original FAA STC.</td>
<td>- Operational approval to use T-PEDs (Transmitting Portable Electronic Devices) must be obtained separately from this STC. - This STC does not constitute installation approval of any cellular or wireless radio frequency systems using wireless data technologies - This STC only to be used concurrently w/...</td>
<td>02/03/2017</td>
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<td>365 AEROSPACE DDS LIMITED</td>
<td>REV. 1</td>
<td>S-76C</td>
<td>Installation of N anyght DAQ6000 as an alternate Flight Data Acquisition Unit (FDAU). Rev. 1: Transfer of STC from 365 Aerospace Ltd. to 365 Aerospace DDS Ltd.</td>
<td>The Certification Basis for the original product as amended by applicable waivers.</td>
<td>This STC requires a pre-approval of the installation of Major Change AESA IM.R.C. 01640, dated 13 September 2004. Prior to installation of this design change it must be determined that the installation will not introduce a risk to the airworthiness of the product.</td>
<td>This STC requires as a prerequisite the installation of Minor Change AESA IM.R.C. 01640, dated 13 September 2004. Prior to installation of this design change it must be determined that the installation will not introduce a risk to the airworthiness of the product.</td>
<td>05/08/2019</td>
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</table>
10081170 365 AEROSPACE 
DOS LIMITED REV. 1 737-800 2024-03-07 737-700 Change to introduce a 158V LOPA and replace installation onto Weber C1000waRev. 1: Transfer of STC from 365 Aerospace Ltd. to 365 Aerospace DOS Ltd

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

10081176 2 EXCEL DESIGN LIMITED REV. 4 A320-211/-212/-214/-215/-216/218 A321-111/-112/-113/-114/-115 A318-111/-112/-121/-122 A319-111/-112/-113 A320-231/-232/-233/-231/-232/-233/251N A320-231/-232/-233/-251N A320-211/-212/-214/-215/-216 A319-111/-112/-133 A319/-111/-112/-113/-114/-115 05/02/2019

10081179 AMES AEROSPACE & MECHANICAL ENGINEERS INC. WALTER STARZACHER Gas A330-200/-202/-203/223/-243 03/07/2021 20/February 2017

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.

10081180 DELTA ENGINEERING CORPORATION REV. 1 FAA STC 737-900ER/9 STO403NY 737-700-800-900 14/02/2022

Installation of a reassess Cabin Connectivity (with demod option) systemRevision 1 expands the installation of the GoGoeVision system to the Boeing 737-900, 737-900ER, 737-800.

10081181 INTELSAT INFLIGHT LLC REV. 1 FAA STO4116CH 06/04/2021

Installation of 802.11 ac protocol Cabin Wireless Access Points project/Waypoint for existing Cabin Wireless Access Points. Rev. 1: Change company name from "GOOG LLC" to "Inmarsat Inflight LLC".

The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.

Installation Instructions as listed on FAA AML Number STO4116CH/Instructions for Continued Airworthiness (ICA) as listed on FAA AML Number STO4116CH.
List of EASA Supplemental Type Certificates

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<tr>
<td>10061190</td>
<td>STENE AVIATION, LLC</td>
<td>REV. 1</td>
<td>FAA STC SA0027LA</td>
<td>DHC-6 SERIES 400 DHC-6 SERIES 300 DHC-6 SERIES 200 DHC-6 SERIES 100</td>
<td>Installation of wing handpumps Rev. 1: Name Change from &quot;Ikhana Group Inc.&quot; to &quot;Ikhana Aviation LLC.&quot;</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Ikhana Aircraft Services Installation Drawing List No. ER218-1230, Revision NC, dated 29 December 2016 or later FAA approved revisions. 2. Ikhana Aircraft Services Master Drawing List No. ER218-1221, Revision NC, dated 29 December 2016 or later FAA approved revisions. 3. Ikhana Aircraft Services Master Drawing List No. ER218-1215, Revision NC, dated 25 December 2016 or later FAA approved revisions. 4. Compliance Check List CCL-34-1604-A Rev. 02 dated 25/02/2017.</td>
<td>09/06/2021</td>
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| 10061210 | AVINCIS AVIATION ITALIA S.P.A | REV. 4 | CL-215-6B11 (CL-415 VARIANT) | | INSTALLATION OF AN INFRARED VISION SYSTEM Rev. 3: Change of STC holder’s name from Babcock Mission Critical Services Italia S.p.A. Divisione Ala Fissa to Avincis Aviation Italia S.P.A., starting from the 1st of March 2023 Rev. 4: Correction of the applicability. | The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement FMS-34-45-1604-A rev. 00 dated 21/10/2016, Master document list MDL-34-45-1604-A rev. 00 dated 25/02/2017. | Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product. | 15/06/2023 | Active | 10061210 |

| 10061220 | STENE AVIATION, INC. | FAA STC SA225WE | SEE FAA APPROVED MODEL LIST | | Installation of leading edge "CUFF" and aileron gap covers on each wing. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | The master drawing list (MDL) as listed on the AML, or later FAA approved revision. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 09/03/2017 | Active | 10061220 |

| 10061221 | STENE AVIATION, INC. | FAA STC SA030WE | | T182, T182T, T182R 182Q, 182R, 182T, 182R 182K, 183L, 182M, 182N, 182P 182E, 182F, 183Q, 182H, 182J 182A, 182B, 182C, 182D | Installation of leading edge "CUFF" on each wing. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Stoner Master Drawing List Number SAS-P-1, Revision A, dated November 20, 2015, or later FAA approved revisions. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 09/03/2017 | Active | 10061221 |

| 10061236 | LANZITZ/PRENA FOLIER FACTORY GmbH | FAA STC SA030WE | | SEE FAA APPROVED MODEL LIST | Application of ORATEX UL600MK3 aircraft covering system within-stitched integration Rev. 2 includes "on-condition" NDE evaluation for the lifetime of the system. | The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS 22.21, CS 22.21 CS22.29, CS 22.45, CS 22.63, CS 22.143, CS22.112, CS 22.201, CS 22.203, CS 22.207, CS 22.251, CS 22.301... | - Application of ORATEX covering & coating system. Doc. No. ADV-51-1-C/001-021-2020-02. | Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ repair will introduce no adverse effect upon the airworthiness of the product. | 29/05/2020 | Active | 10061236 |

| 10061246 | S4A SOLUTIONS FOR AVIATION, S.L. | FAA STC SA030WE | | C-212-CC/C-212-DD | Installation of GPS Repeater. | The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. | Flight Manual Supplement ref. A1206-HDK01 issue 2, dated 23/09/november/2018, or later revisions of the above listed document(s) approved by EASA Certification Plan ref. A1206-CPLO1 issue 12, dated 28 February 2017. | Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 10/03/2017 | Active | 10061246 |
10061248 TURK HAVA YOLLARI TEKNİK A.S. (TURKISH TECHNIC INC.) A320-232 Cabin Retrofit The Certification Basis (CB) for the original product remains applicable to this certificate/approval. Note: no need for the L-3 Communications TDR-94D Enhanced SurveillanceModem S Transponders paired with the Rockwell Collins GPS-4000S WAAS GPS/Inertial, plus installation of the IEC-4000. Master Document CS-25-0008-M001-00 Addressing Compliance Documents in Chapter 2In Part C.2, 3rd revision of the above listed documents approved by EASA. Limited to S/N 2195, 2401Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadenvelope affect upon the airworthiness of the product. 10/03/2017 Active 10661248

10061270 LEARJET Inc. FAA STC ST00610DE CL-600-2B16 (604 VARIANT) Installation of Automatic Dependent Surveillance - Broadcast (ADS-B) Outfunctionality using the Rockwell Collins CIP-310 Enhanced SurveillanceModem S Transponders paired with the Rockwell Collins GPS-4000S WAAS GPS/Inertial, plus installation of the IEC-4000. 1. Bombardier Master Data List, Document No. BAS3000138, Rev. 0, dated 8 October 2016, or later FAA approved revision. 2. Airplane Flight Manual Supplement AFMS, BAS3004023, Rev. IC, dated 28 November 2016 or later FAA approved revision. 3. Instructions for Continued Airworthiness, document BAS 6305061-. 1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadenvelope affect upon the airworthiness of the product. 2. The Input/Output Concentrator installed by... 13/03/2017 Active 10661270

10061274 ASES AEROSPACE & MECHANICAL ENGINEERS INC. WALTER STARZACHER GES A320-233, A330-243 A330-201, A330-202, A330-203 A330-200 Cabin Reconfiguration. The Certification Basis for the original product as amended to the following additional or alternative airworthiness requirements: Special Condition (s): CR- F-01 on Non-Rechargeable Lithium Battery Installations. (The following paragraph(s) at a later amendment: JAR 25.269, change 13, OP 25-93-3. The... Engineering Order A330-200 Cabin Upgrade EO_M170301701_M2_130 62016Rev 0 - Instructions for Continued Airworthiness AICA M170301701 M2_130 62016Rev 1 - Aircraft Flight Manual Supplement AFMS, 17300101, OECD 210 62016 Rev 0 - Document and drawing file A330 Cabin Upgrade 17330085 Rev 3 or later revision... For effective modification prerequisites see Document and Drawing tree A330 Cabin Upgrade 17330085 Rev 3 For effective Compliance modifications see Document and Drawing tree A330 Cabin Upgrade 17330085 Rev 3 Note: Drawing Tree A330 Cabin Upgrade 17330085 Rev 3 include有效性 list of affected air... 13/03/2017 Active 10661274

10061277 TURK HAVA YOLLARI TEKNİK A.S. (TURKISH TECHNIC INC.) A320-232 A321-231 A319-132 Installation of wireless IFE System. The Certification Basis for the original product as amended to the following additional or alternative airworthiness requirements: Special Condition (s): Aircraft Flight Manual Supplement CS002-AFM-00 or later FAA approved revision of the above listed documents approved by EASA Certification Programme C0002-CERP-03. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadenvelope affect upon the airworthiness of the product... 13/03/2017 Active 10661277

10061283 ADS-B TECHNOLOGIES, L.L.C. FAA STC ST00798AR SEE FAA AMI Installation of the I-3 Communications Display System EF1450. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. New-PAGE The requirements for environmental protection and the associated noise level’s of the original product are unchanged and remain applicable to this certificate/approval. 1. Master Data List L3-EFI-650-AMS, Revision 0, dated 20 September 2016 or later FAA approved revision. 2. Instructions for Continued Airworthiness L3-EFI-650ICA, Revision 0, dated 16 October 2016, or later FAA approved revision. 1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadenvelope affect upon the airworthiness of the product... 16/05/2017 Active 10661283

10061304 TENENCIA Ltd. CAA UK RESPONSIBILITY TENENCIA Ltd. A321-213/231/-232 A321-111/-112/-121/-211/- 212 A320-231/-232/-233 A320-211/212/214/-215/- 216 A319-111/130/-133 A319-111/112/-113/-114/- 115 ARINC 791 SYSTEM PROVISIONS. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise level’s of the original product are unchanged and remain applicable to this certificate/approval. TENENCIA LTD. Flight Manual Supplement No. TEN-008-47 Issue 1. TENENCIA LTD. INSTRUCTIONS FOR CONTINUED AIRWORTHINESS No. TEN-008-04/2014 and/or later revisions of the above listed documents approved by EASA-TENENCIA LTD. MASTER DATA LIST No. TEN-008 Issue 1 dated 10 March 2017. None. 15/03/2017 Active 10661304
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<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
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<th>Associated Technical Documentation</th>
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<td>10061309</td>
<td>AEROTEC &amp; CONCEPT</td>
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<td>767-8, 767-0</td>
<td>B777 DUAL MEDICAL STRETCHER INSTALLATION.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate’s approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>- Master Data List AE16230MIL Rev.0 dated 10/33/2017, FAA Supplement AE16230AFMS Rev.0 dated 08/1/2017, Instruction for Continued Airworthiness AE16230CAA Rev.0 dated 07/1/2017, or later versions of the above listed documents approved by EASA.</td>
<td>Per AFM Supplement AE16230AFMS - Aerotec Change AE16230AFMS is a prerequisite for this STC. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. The STC is valid for all aircraft types...</td>
<td>16/02/2017</td>
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<td>10061316</td>
<td>APICAL INDUSTRIES, Inc./bioa DART AEROSPACE</td>
<td>FAA STC</td>
<td>SR02286LA</td>
<td>A842/ A8412EP</td>
<td>Installation of Apical Industries, Inc. Cable Cutters Kit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate’s approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>1. FAA approved Apical Industries, Inc. Master Document List MD412-101, Revision A, dated 23 December 2009, or later FAA-approved revision.2. FAA-approved revision.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. The STC is valid for all aircraft types...</td>
<td>22/03/2017</td>
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<td>10061321</td>
<td>STANDARD AERO BUSINESS AVIATIONSERVIC ES, LLC</td>
<td>REV. 2</td>
<td>FAA STC</td>
<td>ST1112SC</td>
<td>A320-232</td>
<td>Installation of a VIP interior.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s)/Condition(s) 2302-002 (Isolated compartments)/CRI D-2306-001 (Application of heat release and smoke density requirement to seat material)- Deviation(s): CRI D-2306-001.</td>
<td>1. Master Document List, Associated Air Center document No. 3260400000, Revision C dated 13 October 2017.2. Instruction for Continued Airworthiness, Associated Air Center document No. 3260400000, Revision D dated 18 August 2017.3. Aircraft Flight Manual Supplement, Associated Air Center document No. 3...</td>
<td>The Cabin Wireless Local Area Network (WLAN) system is intended to provide internet connection and e-mail services using portable electronic devices (PEDs). Any other intended function of this equipment will require a re-examination. Components identified in AAC document Command off the Shelf PMA(Ex)…</td>
<td>13/02/2018</td>
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<tr>
<td>10061325</td>
<td>AVDYN E CORPORATION</td>
<td>FAA STC</td>
<td>SA00352BO</td>
<td>SEE FAA AML</td>
<td>Installation of an Avdyne Corporation Mode AC/CS Transponder with ADS-Bout capability. Model APX340 loaded with Version 3.13 software (or later FAA approved version) (or Model APX322 with Version 2.7 software (or later FAA approved version).</td>
<td>The Certification Basis for the original product remains applicable to this certificate’s approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>1. Avdyne Corporation Master Document List (MDL), Document Number AVXP0-000, Revision 0 dated 9 January 2015, or later FAA approved revision.2. Avdyne Corporation Aircraft Flight Manual Supplement (AFMS), Document Number 600-00309-000, Revision 01, or later FAA approved revision.3. Instructions for Continued Airworthiness of the above listed documents approved by EASA.4. Further instructions for Continued Airworthiness of the above listed documents approved by EASA.</td>
<td>1. Model APX340 can interface only with proven compatible avionics systems as identified by vendor name and model number in Avdyne Corporation Installation Manual Section 6.8.4, Document Number 600-00358-000, Revision 00, dated 9 January 2015, or later FAA approved revision2. Model APX340 must have aK...</td>
<td>01/08/2017</td>
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<td>10061341</td>
<td>HELI AUTR I A GmbH</td>
<td>REV. 1</td>
<td>AS 332 L, AS 332 L1</td>
<td>SEE FAA AML</td>
<td>Installation of GARMIN GTN750 COM/ NAV/ GPS System into Airbus Helicopter AS332L and AS332L1.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements Special Condition (s)/Condition(s) 62 Non rechargeable Lithium Battery CS-ACNS.B.V.CS.001, CS-ACNS.B.V.CS.010, CS-ACNS.B.V.CS.020 CS-ACNS.B.V.CS.025, CS-ACNS.B.V.CS.040 The require...</td>
<td>Flight Manual Supplement, Doc. No. HAU-D-029-005-05, Rev. 1Flight Manual Supplement, Doc. No. HAU-D-029-005-05, Rev. 2. Instruction for Continued Airworthiness of the above listed documents approved by EASA.1. Flight Manual Supplement, Doc. No. HAU-D-029-005-05, Rev. 2.2. Instruction for Continued Airworthiness of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. The STC is valid for all aircraft types...</td>
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<td>10061346</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>REV. 2</td>
<td>737-300</td>
<td>S-TEC</td>
<td>Interior Outfitting</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-02 Isolated Compartment(s): CRI D-06 Glass Screens of Displays / Monitors (s): CRI D-09 Shower installation (s): CRI D-20 Installation of Glass Panels in the C.</td>
<td>Master Data List MDL-FE01-2500 Rev. L dated 27/04/2018 Airplane Flight Manual Supplement AFMS- FE01-2500 Rev. D dated 12/04/2018.</td>
<td>For aircraft operations with Cabin Crew Limited to MSN 620099 (FAA STC 70000/0/00) Per ICA-FE01-2500 for type approval.</td>
<td>07/05/2018</td>
<td>Active</td>
<td>10061346</td>
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<tr>
<td>10061347</td>
<td>CORPORATION LOGIC AIR Inc</td>
<td>REV. 1</td>
<td>TCCA SA15-34</td>
<td>737-300, 737-400, 737-500, 737-300 / 737-400</td>
<td>Installation of Second Flight Management Computer (FMC) (Revision 1 includes 737-500 model)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Modification Data Summary – Doc Ref: K403000 Rev. D Constructions for Continued Airworthiness – Doc Ref: MM-34354-1 Rev. 1 for later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU Canada Bilateral Agreement.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>23/10/2020</td>
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<td>10061349</td>
<td>BELL HELICOPTER TEXTRON, Inc</td>
<td>FAA STC</td>
<td>70884SW</td>
<td>212/ 412/ 412EP</td>
<td>Installation of Radome assembly.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Aeronautical Accessories, Inc. Drawing 229P796+ (&gt;;&lt;) Rev. A(+;&gt;); dated 26 March 1988. Servo Mount installation in accordance with Aeronautical Accessories, Inc. Drawing List Report AA-92020, dated 27 April 1992 or FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>20/02/2017</td>
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<td>10061350</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC</td>
<td>SA0020-14C-D</td>
<td>182S</td>
<td>Installation of S-TEC System SS/15X Two Axes Automatic Flight Guidance System, Model ST-686</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Bulletin No. 666. Revision 2, dated 2 February 2001. 2. Master Flight Manual Supplement LS-000 Rev. B dated 2 February 2001 (Pilot’s Operating HandBook). 3. FAA/DAS Approved Pilot’s Operating Handbook – FAA/DAS Approved Pilot’s Operating Handbook – FAA/DAS Approved Pilot’s Operating Handbook – FAA/DAS Approved Pilot’s Operating Handbook.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/03/2017</td>
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List of EASA Supplemental Type Certificates

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<td>ASI AVIATION</td>
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<td></td>
<td>B300 AND B300C</td>
<td>Installation, operation and maintenance of ASI Aviation design change “installation of VHF Marine” in accordance with DL, Data List for FAM618 No. 027 issue 00, dated 28 October 2017. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- DL - Data List for FAM 518 No. 027 issue 00, dated 28 October 2017; Aircraft Maintenance Manual Supplement MM-350-95000.00 issue 00, dated 21 December 2018; or later revisions of the above listed documents approved by EASA. - Flight Manual Supplement STC-107-0-FIM, Rev. 0 for later revisions of the above listed documents approved by EASA. - EASA - Maintenance Manual Supplement STC-107-0-AMM, Rev. 0. Installation Instruction STC-107-6-6, Rev. 0.</td>
<td>Prior to installation of this design change it must be determined that the interoperability between this design change and any previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/02/2017</td>
<td>Active</td>
<td>10061420</td>
<td></td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td></td>
<td></td>
<td>EC135 P2+</td>
<td>Operator Seat Installation (Fischer Seat 170/380 H160) The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: Amendment 2 of CS-27 has been chosen for the following paragraphs: CS27.21, CS27.23, CS27.301, CS27.303, CS27.305, CS27.307.</td>
<td>- Master Data List MOD00045_MDL Rev. 3 dated 27/03/2017; Aircraft Flight Manual Supplement MOD00045_AFM Rev. 3 dated 06/03/2017; Instructions for Continued Airworthiness MOD00045_ICA Rev. 1 dated 01/11/2015 for later revisions of the above listed documents approved by EASA. - EASA - A330 MRTT Cabin Removal Certification Plan Ref. ACJC-AJA-16209. - ACJC Modification Approval Sheet Ref. C.J007 issue 1 dated 19.03.2017.</td>
<td>Prior to installation of this design change it must be determined that the interoperability between this design change and any other previously installed product remains unchanged.</td>
<td>28/03/2017</td>
<td>Active</td>
<td>10061436</td>
<td></td>
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<tr>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td></td>
<td></td>
<td>DHC-8-315 DHC-8-102</td>
<td>Installation of MNL527 Airborne Stretch on DHC-8 Series The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-01 MEDEVAC/SA02267LA The following paragraph(s) at a later amendment: Amendment 2 of CS-27 has been chosen for the following paragraphs: CS27.21, CS27.23, CS27.301, CS27.303, CS27.305, CS27.307.</td>
<td>- Master Data List MOD00045_MDL Rev. 3 dated 27/03/2017; Aircraft Flight Manual Supplement MOD00045_AFM Rev. 3 dated 06/03/2017; Instructions for Continued Airworthiness MOD00045_ICA Rev. 1 dated 01/11/2015 for later revisions of the above listed documents approved by EASA. - EASA - A330 MRTT Cabin Removal Certification Plan Ref. ACJC-AJA-16209. - ACJC Modification Approval Sheet Ref. C.J007 issue 1 dated 19.03.2017.</td>
<td>Prior to installation of this design change it must be determined that the interoperability between this design change and any other previously installed design change remains unchanged and no adverse effect will be introduced upon the airworthiness of the product.</td>
<td>28/03/2017</td>
<td>Active</td>
<td>10061437</td>
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<tr>
<td>AIRBUS S.A.S.</td>
<td></td>
<td></td>
<td>A330-203</td>
<td>Cabin Removal for MRRT Conversion The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI DJ-01 MEDEVAC/SA02267LA The following paragraph(s) at a later amendment: Amendment 2 of CS-27 has been chosen for the following paragraphs: CS27.21, CS27.23, CS27.301, CS27.303, CS27.305, CS27.307.</td>
<td>- Master Data List MOD00045_MDL Rev. 3 dated 27/03/2017; Aircraft Flight Manual Supplement MOD00045_AFM Rev. 3 dated 06/03/2017; Instructions for Continued Airworthiness MOD00045_ICA Rev. 1 dated 01/11/2015 for later revisions of the above listed documents approved by EASA. - EASA - A320 MRTT Cabin Removal Certification Plan Ref. ACJC-AJA-16209. - ACJC Modification Approval Sheet Ref. C.J007 issue 1 dated 19.03.2017.</td>
<td>Prior to installation of this design change it must be determined that the interoperability between this design change and any other previously installed product remains unchanged.</td>
<td>28/03/2017</td>
<td>Active</td>
<td>10061444</td>
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<tr>
<td>IKhANA GROUP, LLC</td>
<td>REV. 3</td>
<td>SA02267LA</td>
<td>DHC-8-300 DHC-8-400</td>
<td>This STC approves the Viking Air Limited DHC-8-300 and DHC-8-400/402 aircraft to take off and land at a maximum weight of 14,000 pounds and DHC-8-300 and DHC-8-400/402 aircraft to take off and land at a maximum weight of 14,000 pounds when operated under Restricted Category for theoretical purposes. The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. R. W. Martin Master Data List No. ER218-3041, Revision A, dated 12 November 2009, or later FAA approved revisions, and if applicable, R. W. Martin Master Data List No. 218-3051, Revision NC, dated 5 July 2010, or later FAA approved revisions (DHC-8-300 and DHC-8-400 multiplace) takeoff at a maximum weight of 12,500 pounds must be operated under Restricted Category 3. Prior to installation of this design change it must be determined that the interoperability between this design change and all other previously installed design change and/or repair will be inoffensive upon the airworthiness of the aircraft.</td>
<td>08/09/2021</td>
<td>Active</td>
<td>10061445</td>
<td></td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>REV. 1</td>
<td>DORNIER 228-212</td>
<td>Approval of EFIS installation &amp; Avionics Upgrade. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI B-52 Human Factors CRI B-52 HRF CRI F-4 Lightning Indirect Effect</td>
<td>Installation in accordance with Scandinavian Avionics Aircraft Flight Manual Supplement: STC-2102 491-AM.A EFS &amp; E.</td>
<td>Prior to installation of this design change it must be determined that the interoperability between this design change and any other previously installed design change and/or repair will be inoffensive upon the airworthiness of the product.</td>
<td>30/03/2017</td>
<td>Active</td>
<td>10061450</td>
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<tr>
<th>STC Holder</th>
<th>EASA STC No.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>AIRBUS S.A.S.</td>
<td>10061456</td>
<td>A330-203</td>
<td>CABIN INSTALLATION FOR MRRT CONVERSION.</td>
<td>- AC20.130K Modification Approval Sheet Ref. C32720 dated 10.03.2017, A330MRRT Cabin Installation Certification Plan Ref. ACUC-AJAA-1030.</td>
<td>1. STC is limited for installation on MSN 0892 and 0898.2. Installation of this STC is in conjunction with EASA STC 10061444.3. Zero occupancy and no cargo allowed in the passenger cabin during TTOL and flight. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/03/2017</td>
<td>Active</td>
<td>10061456</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>10061459</td>
<td>767-300</td>
<td>KA BAND Antenna installation.</td>
<td>- MASTER DATA LIST Dsc. No. AE10293ML Rev.0 dated 29.02.2017 INSTRUCTION S FOR CONTINUED AIRWORTHINESS DESIG N „AE10293ICA Rev.1 „AE10293ICA Rev.1 dated02/2017 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/03/2017</td>
<td>Active</td>
<td>10061459</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>10061462</td>
<td>EC135 P2+</td>
<td>TRAKKA A 800 Searchlight Installation</td>
<td>-STC-106-0-AMM, Rev.1, dated 15-March-2017, AMM Supplementor later revisions of the above listed documents approved byEASA- STC-106-0-AMM, Revision 1, dated 03-07-2017, Maintenance ManualSupplement. STC-106-0-6, Revision 2, dated 15-March-2017, Installation Instructions- STC-AT-108, Revision 3, 24-Mar-...</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/03/2017</td>
<td>Active</td>
<td>10061462</td>
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<tr>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>10061463</td>
<td>737-700</td>
<td>Installation of passive passenger oxygen system</td>
<td>-Master Data List ML- B737-3/500 Rev. B dated 29.03.2017. Airplane Flight Manual Supplement AFMS-B737-3/500 Rev. B dtd.23-Mar-2017, Instructions for Continued Airworthiness ICA- B737-3/500 Rev. C dtd.28-Mar-2017 or later revisions of the above listed documents approved by EASA.</td>
<td>Limited to MSN 6269PPer AFMS-B737-3/500Per ICA- B737-3/500Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the...</td>
<td>31/03/2017</td>
<td>Active</td>
<td>10061463</td>
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<tr>
<td>CESSNA DUESSELDORF CITATIONSERVICE CENTER GmbH</td>
<td>10061473</td>
<td>525, 525A</td>
<td>ADS-B upgrade of Garmin GTX-3300.</td>
<td>- Airplane Flight Manual Supplement 2017-2-6, Amdt.50 later revisions of the above listed documents approved by EASA Master Documents List 2017-2-16, Amdt. IR- Engineering Order 2017-2, Amdt. IR- Maintenance Manual Supplement (inc. Instructions for Continued Airworthiness) 2017-2-10, Amdt. IR-...</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/03/2017</td>
<td>Active</td>
<td>10061473</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC no.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10061475</td>
<td>MAGELLAN SERVICES COMPANY</td>
<td>REV. 1</td>
<td>FAA STC ST110685C</td>
<td>737-700 737-800/737-900/8</td>
<td>Installation of a dual port heated drain mast assembly Revision 1: Additional model in the applicability list</td>
<td>The Certification Basis for the original product as amended including additional or alternative airworthiness requirements CS 25 at amendment 11; 25.613, 25.1360, The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged.</td>
<td>Magellan Services Company Master Document List 1002000000, Rev. E, dated 7 November 2012, or later FAA approved revision. Instructions for Continued Airworthiness, Magellan Services Company Document Number 100205090, Rev. A, dated 21 November 2011 or later FAA accepted revision.</td>
<td>09/02/2020</td>
<td>Active</td>
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<tr>
<td>10061476</td>
<td>MT-PROPPELLER ENTWICKLUNG GmbH</td>
<td></td>
<td>A-1B, A-1C-180, A-1, A-1A</td>
<td>MTV-6/200-50</td>
<td></td>
<td>The Certification Basis for the original product as amended including additional or alternative airworthiness requirements, the following paragraph(s) at a later amendment: CS 36, Amendment 4, FAR 36, Amendment 36-307b this certificate's approval involves a change to the requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged.</td>
<td>According to MT-Propeller Master Documentation List No. E-2845, Rev 07c be operated in accordance with MT-Propeller AFM-S Document No. E-2822, Rev 0 or later revisions of the above listed documents approved by EASA. To be installed i.a.e. Installation Instruction Doc. No. E-2653, Rev 0 or later Revision...</td>
<td>03/04/2017</td>
<td>Active</td>
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<tr>
<td>10061477</td>
<td>TELEDYNE CONTROLS, LLC</td>
<td>FAA STC ST00226A</td>
<td>ERJ 190-200 STD ERJ 190-200 LG ERJ 190-100 LG ERJ 190-100 GD-ERJ 190-100 ECJ 685-600</td>
<td>Installation of Teledyne Controls Wireless Ground Link - Quick Access Recorder (WQAR)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate's approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate's approval.</td>
<td>Teledyne Controls Master Data List Document No. MDL 2243800-E190-Revision 5*, dated 30 August 2016, or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>06/10/2017</td>
<td>Active</td>
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<td>10061480</td>
<td>EMBRAER S.A.</td>
<td>REV. 1</td>
<td>ANAC PROCESS H. 02-4766-0</td>
<td>EMB-500</td>
<td>MEPT CONFIGURATION (MULTI-ENGINE PILOT TRAINING)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate's approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate's approval.</td>
<td>To be installed in accordance with documents listed in Master Document List ESCCD0063 rev. E dated 9th Jan 2017 to be operated in accordance with AFM Supplement H. 02-4766-0/EASA/2017rev.1.P.R dated 22nd March 2017. To be maintained in accordance with Instructions for Continued Airworthiness ESOEMD0341 rev. S.</td>
<td>None</td>
<td>06/04/2017</td>
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<tr>
<td>10061481</td>
<td>DAO AVIATION A/S</td>
<td>REV. 1</td>
<td>B200</td>
<td>Installation of Garmin Radar GWX70 interfaced with GNT750</td>
<td>See appendix of TV The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate's approval.</td>
<td>Master Data List DAO-DO-0569-MLD-00 dated 21-03-2017 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>10/04/2017</td>
<td>Active</td>
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<tr>
<td>10061484</td>
<td>AVIATION TRADERS LIMITED/EUROPE S.A.S. REV. 1</td>
<td>767-300</td>
<td>Installation of Artex C406-2 Fixed ELT Rev. 1 Transfer of STC</td>
<td></td>
<td>The Type and OSD Certification Basis for the original product as amended including additional or alternative airworthiness requirements. Special Condition(s): The following(s) Special Condition(s) applicable to this change: CRI-7010 Non-Rechargeable Lithium Battery Installation. The following pa...</td>
<td>- 13065-000001, Issue 1, Flight Manual Supplement, Installation of ARTEX Model C406-2 Fixed ELT or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>10/11/2021</td>
<td>Active</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA Doc. No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10061493</td>
<td>HARTZELL PROPELLER INC.</td>
<td>Active</td>
<td>FAA STC SA0456CH</td>
<td>PA-46-500TP (MERIDIAN)</td>
<td>Installation of 5-blade Raptor propeller installation of Hartzell propeller model SD1-N33A176D1-18 or PA-46-500TP, SN 4697001 and up</td>
<td>-AFMS 020215, Initial Release (IR)</td>
<td></td>
<td>S/N 4697001 and up</td>
<td>04/04/2017</td>
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<td>10061495</td>
<td>ANDRA INTERNATIONAL LLC</td>
<td>Active</td>
<td>FAA STC ST02502LA</td>
<td>CL-600-2B19 (RJ SERIES 100)</td>
<td>Installation of main deck cargo loading system.</td>
<td>1. Angra International LLC Controlled Data List CL/5450 Revision Dated 4 May 2016, or later FAA approved revision. 2. Instructions for Continued Airworthiness Cargo Handling System. Part Numbers 85400-10 and 85400-11 installed on CR200 (CL-600-2B19) freighter.</td>
<td></td>
<td></td>
<td>07/06/2018</td>
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<tr>
<td>10061529</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>Active</td>
<td>CAA UK RESPONSIBIL</td>
<td>AS 355 NP</td>
<td>AutoPilot HDG mode-GPS Coupling with GTN650 as per change AHUK/355/3073.</td>
<td>AS355 FMS SUPPLEMENT No. 103, AUTOPILOT GPS COUPLING WITH GTN650/AHUK/355-FMS-103, Rev. C, April 2017 or later revisions of the above listed documents approved by EASA for continued airworthiness, ICA/255073-01, Rev. A, 14th December 2016 (No ALS identified for this STC) Design Compliance.</td>
<td>This STC is compatible only with Autopilot Configuration AS355FMS-01, and AS355FMS-16, (AFCS EFIS IFI T 31, 3-axis, with optional coupler) and AHUK SPP: 355-FMS-101 (Installation of GTN650 as primary COM/NAV). Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design changes and/or repair will introduce any adverse effect upon the airworthiness of the product.</td>
<td></td>
<td>07/04/2017</td>
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<tr>
<td>10061532</td>
<td>BOMBARDIER Inc.</td>
<td>Active</td>
<td>TCCA SA02-131 ISSUE 3</td>
<td>BD-700-1A10</td>
<td>ESA validation of Transport Canada Supplemental Type Certificates SA02-131 at issue 3 Upgrade to Honeywell MCS-7000 SATCOM</td>
<td>Installation of Honeywell MCS-6000 SATCOM System in accordance with Bombardier Service Bulletin SB700-23-011 is a pre-requisite for this modification. Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed...</td>
<td></td>
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<td>10/04/2017</td>
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<td>STC Holder</td>
<td>FAA STC Holders</td>
<td>STC No.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td><strong>GULFSTREAM AEROSPACE CORPORATION</strong></td>
<td></td>
<td><strong>STC10760SC-D</strong></td>
<td>Installation of an Aircraft Axxess II Airborne Satcom System with an Airline Cabin Telecommunications Router (CTR) -- OFF -- an Airline Cabin Telecommunications Router (CTR) ONLY -- OFF -- an Airline Air to Ground (ATG) Communications System in combination with either the Axxess II Airborne Satcom System...</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Gulfstream Index List GA212038002, Rev Q, dated 15 July 2016</td>
<td>Prior to installation of this design change it must be determined that the installation will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/04/2017</td>
<td>Active</td>
<td><strong>10061532</strong></td>
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<tr>
<td><strong>GULFSTREAM AEROSPACE CORPORATION</strong></td>
<td></td>
<td><strong>STC10761SC-D</strong></td>
<td>Installation of a Thrane and Thrane Aero SB Lite High Speed Data System</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Gulfstream Index List GA212038001, Rev -, dated 14 August 2009 - required for the Aircraft Axxess II Satcom System Flight M...</td>
<td>STC STC10760SC-D is required prior to or concurrently with this installation. Prior to installation of this design change it must be determined that the installation will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/04/2017</td>
<td>Active</td>
<td><strong>10061534</strong></td>
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<tr>
<td><strong>AVONIK STRAUBING ENTWICKLUNGS GmbH</strong></td>
<td></td>
<td><strong>REV. 1</strong></td>
<td>Installation of Garmin G500 system.</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment: CS-25, Amendment 423, 251, 301, 305, 307, 309, 317, 361, 603, 605, 607, 609, 611, 613, 771, 773, 777, 863, 867, 1301, 1303...</td>
<td>Flight Manual Supplement ARS-2015-115-FMS-05-00, Rev. 00 - later revisions of the above listed documents approved by EASA</td>
<td>A suitable Garmin GPS System must have been already installed. This STC includes the installation of Garmin’s Synthetic Vision (SV) Option (Software). Prior to installation of this design change it must be determined that the installation will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/04/2017</td>
<td>Active</td>
<td><strong>10061535</strong></td>
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<tr>
<td><strong>ASI AVIATION</strong></td>
<td></td>
<td><strong>B300</strong></td>
<td>POLMAR mission equipment installation.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Aircraft Flight Manual Supplement - Document n°: POH-350-958000 Issue00, dated 09 March 2017</td>
<td>None.</td>
<td>10/04/2017</td>
<td>Active</td>
<td><strong>10061536</strong></td>
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<td><strong>AEROSERV, S.R.O.</strong></td>
<td></td>
<td><strong>IL 410 UVP-E / IL 410 UVP-E20</strong></td>
<td>Installation of Chelton IDU-450 EFIS.</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment: CS-25, Amendment 423, 251, 301, 305, 307, 309, 317, 361, 603, 605, 607, 609, 611, 613, 771, 773, 777, 863, 867, 1301, 1303...</td>
<td>Certification program AE 118/16 P Rev. H.The Flight Manual Supplement/L 410 UVP-E FLIGHT MANUAL SUPPLEMENT, Nov 05/16, STC AE118-AEP-115-L 410 UVP-E20 FLIGHT MANUAL SUPPLEMENT, Nov 05/16, STC AE118-AEP-115-L 410 UVP-E20 FLIGHT MANUAL SUPPLEMENT, Nov 05/16, STC AE118-AEP-115-L 410 UVP-E20 FLIGHT MANUAL SUPPLEMENT, Nov 05/16, STC AE118-AEP-115-L 410 UVP-E20 FLIGHT MANUAL SUPPLEMENT, Nov 05/16...</td>
<td>None.</td>
<td>12/04/2017</td>
<td>Active</td>
<td><strong>10061569</strong></td>
</tr>
</tbody>
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**Notes:**
- The list includes supplemental type certificates (STCs) issued by the European Union Aviation Safety Agency (EASA).
- The documents referenced include FAA STC Holders and relevant models, descriptions, certification basis, associated technical documentation, limitations/conditions, and dates of issue.
- The list is for STC ST10760SC-D.
<table>
<thead>
<tr>
<th>STC Holder</th>
<th>FAA STC</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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</thead>
<tbody>
<tr>
<td>SHADIN AVIONICS</td>
<td>FAA STC SR04158Ch</td>
<td>Installation of a fuel flow adapter.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Prior to installation of this design change it must be determined the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/07/2017</td>
<td>Active</td>
<td>10061590</td>
</tr>
<tr>
<td>AEROMECH Inc.</td>
<td>FAA STC ST05096E</td>
<td>Installation of dual Garmin GTX 3000 Transponders with ADS-B Out.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Aircraft must be modified with dual GTX 725 GPS Navigator System per Garmin EASA STC 150/482/12. Aircraft must be modified with ACSS RCZ 852 Transponders. per EASA STC 150/482/12.</td>
<td>21/04/2017</td>
<td>Active</td>
<td>10061588</td>
</tr>
<tr>
<td>STAR AVIATION, Inc.</td>
<td>FAA STC ST03726Y</td>
<td>Installation and modification upgrade of the existing Cabin Wireless Access Point (CWAP), SMART monitor and file server</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: 25.850(a)(4), 25.869(a)(3) and 25.896(b)(1)(2) at CS25-Amdt.0. The requirements for environmental protection and the associated certification do not apply.</td>
<td>- Airplane Flight Manual Supplement. Ref: R44-1439, Rev. C - Instructions for Continued Airworthiness. Ref: R44-1405, Rev. C - The later revisions of the listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions other than this decision) and/or the Technical Documentation.</td>
<td>- The following STCs are prerequisites to this installation: - EASA STC Ref. EASA.IM.A.S. 00285 - EASA STC Ref. 1000/4148 - Prior to activation of a CWAP, it must be shown that the aircraft is compatible with transmittable portable electronic devices (T-PED). - No design changes to required.</td>
<td>19/04/2017</td>
<td>Active</td>
<td>10061589</td>
</tr>
<tr>
<td>PMV ENGINEERING</td>
<td>REV. 3</td>
<td>TPED tolerance on 8777. The change consists of introducing the updated ED-130A/DG-360 and using the EASA CS-ES-003 (PED tolerance) instead of CRJ WLAN and TGL29.</td>
<td>The Certification Basis for the original product is amended as follows: The following paragraph(s) at a later amendment: 25.1303(a)(x)(c), 25.1303(a)(x)(j), 25.1303(a)(x)(k), 25.1303(a)(x)(l) and 25.1528 at CS25-Amdt.0. The requirements for environmental protection.</td>
<td>- MASTER DOCUMENT LIST, ref: SCA-00800-MDL-R05, dated 27/02/2011: INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, ref: SCA-00800-ICA-R02, dated 27/02/2011: INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, ref: SCA-00800-ICA-R02, dated 27/02/2011: AIRCRAFT FLIGHT MANUAL SUPPLEMENT, ref: SCA-00800-AFM-R00, etc.</td>
<td>Per applicable AFM Supplement SCA-00800-AFMS. Per applicable instructions for Continued Airworthiness SCA-00800-ICA. Prior to installation of this design change it must be determined whether the relationship between this design change and any other previously installed design change and/or repair will...</td>
<td>06/03/2018</td>
<td>Active</td>
<td>10061590</td>
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<tr>
<td>SPECTRUM AEROMED, LLC dba SPECTRUM AEROMED</td>
<td>REV. 1 FAA STC SR02666Ch</td>
<td>Installation of air ambulance conversion EASA STC 10061651 from SNAKEY CREEK ENTERPRISES, LLC dba SPECTRUM AEROMED to SPECTRUM/AEROMED.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Prior to installation of this design change it must be determined whether the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/04/2019</td>
<td>Active</td>
<td>10061591</td>
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<tr>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AERONAUTICAL ENGINEERS Inc.</td>
<td>CRJ-200Passenger to Freighter Conversion: CRJ-200 Passenger to Freighter Conversion. This modification adds larger cargo door forward of the wing, 9G barrier per Ventura Aerospace LLC FAA-STC No. ST02607LA, Class E cargo compartment, and cargo handling system by Avionics International LLC FAA-STC No. ST026...</td>
<td>AES-767-4135</td>
<td><strong>Installation of Internal Wing Hardpoints:</strong>&lt;br&gt;The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Prior to installation of this change/repair it must be determined that...</td>
<td>07/05/2018</td>
<td>Active</td>
<td>10061615</td>
<td></td>
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<tr>
<td>AVCON INDUSTRIES, Inc.</td>
<td>10061623 55, 55B, 55C, 60 31, 31A, 35, 38, 38A, 25, 25A, 25B, 25C, 25D, 25F Installation of Internal Wing Hardpoints.</td>
<td><strong>Certification Basis</strong> (CB) for the original product as amended by this STC approval. <strong>Certification Basis</strong> (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td><strong>EASA</strong> Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td><strong>Introduction of a Twin Class Cabin Configuration:</strong> Revision 1 removes the MSN limitation. Revision 2 corrects the limitations. Revision 3 transfers from AES Ltd. to new STC holder AESGlobal Ltd.</td>
<td>28/04/2021</td>
<td>Active</td>
<td>10061628</td>
<td></td>
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<tr>
<td>DOA AESGLOBAL LIMITED</td>
<td>10061628 767-300</td>
<td><strong>Certification Basis</strong> (CB) for the original product as amended by this STC approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td><strong>AES</strong> Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td><strong>The following modifications require endorsement prior to or concurrently with this Major change:</strong> Modification AES-767-3767 (FAA STC No. 10061629) - Introduction of a Twin Class Digecor ‘Glide’ IFE System - B767 - 2. Modification AES-767-4135 - Activation of a Twin Class Digecor ‘Glide’ IFE System - B767 - 2.</td>
<td>28/04/2021</td>
<td>Active</td>
<td>10061629</td>
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<tr>
<td>STC TWENTY ONE LIMITED</td>
<td>10061630 CL-600-2B19 840</td>
<td><strong>Certification Basis</strong> (CB) for the original product as amended by this STC approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td><strong>AES</strong> Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td><strong>The additional installation of a rigid main deck cargo barrier per EASASTIC 1004678 (FAA STC ST02607LA) and cargo handling system per EASA STC10061499 (FAA STC ST02601A) which is compatible with this STC is unrequired. Prior to installation of this change/repair it must be determined that...</strong></td>
<td>21/04/2017</td>
<td>Active</td>
<td>10061630</td>
<td></td>
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<tr>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>10061636 TCCA</td>
<td><strong>Certification Basis</strong> (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td><strong>AES</strong> Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td><strong>Prior to installation of this Major change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</strong></td>
<td>21/04/2017</td>
<td>Active</td>
<td>10061636</td>
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<tr>
<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>DHC-8-311/DHC-8-315/ DHC-8-301/DHC-8-820/ DHC-8-302/ DHC-8-102/DHC-8-103/ DHC-8-108</td>
<td>Passenger Briefing System Installation.</td>
<td>Configuration TCD Passenger Briefing System 92023-401, Revision 5, dated 06 October 2008, Engineering Order No. 92023-701 revision 2, dated 06 October 2008, Maintenance Manual Supplement.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/04/2017</td>
<td>Active</td>
<td>10061632</td>
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<tr>
<td>FAA STC SA1406CH</td>
<td>SEE FAA AML</td>
<td>Installation of an air ambulance conversion kit.</td>
<td>1. Med-Pac, Inc. Drawings as listed on FAA AML SA01496CH issued 26 July 2001, or later FAA approved revisions 2. FAA approved Aircrane Flight Manual Supplement approved 26 July 2001, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/04/2017</td>
<td>Active</td>
<td>10061639</td>
<td></td>
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<tr>
<td>Avionik Straubing Entwicklungs GmbH</td>
<td>DA 40 D</td>
<td>Garmin GTN650 Dual Installation</td>
<td>As defined in the TCD of EASA A.022, plus CS-ACNS Initial Issue, Subpart E. Section 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation, Operation and Maintenance according to the Avionik Straubing “Master Document List” - ASR-2016-002-MDL 01-00 Initial Release dated 20/04/2017, or later revisions of the above listed documents approved by EASA.</td>
<td>25/04/2017</td>
<td>Active</td>
<td>10061651</td>
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</tr>
<tr>
<td>SWS Certification Services Ltd.</td>
<td>MBB-BK117 C-1 MBB-BK117 B-2 MBB-BK117 A-4 MBB-BK117 A-3</td>
<td>High Density (9 Pax) Seating Installation On MBB-BK117 Rotorcraft</td>
<td>3159-FMS-01 issue 3, dated 16 April 2017, “Flight Manual Supplement” or later revisions of the above listed documents approved by EASA. EASA 3159-MMS-01, issue 3, dated 24 April 2017, Maintenance Manual Supplement” (no ALS identified for this STC) 3159-COM-01, issue 2 dated March 2017, “Compliance Report”...</td>
<td>Limited to DA-40D aircraft with G500 and/or analog HSI/CDI. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/04/2017</td>
<td>Active</td>
<td>10061652</td>
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<tr>
<td>PAN Engineering</td>
<td>A321-211/-212/-213/-215/- 232 A321-115/-112/-131 A320-211/-212/-213/-215/- 216 A319-131/-120/-133 A320-211/-212/-113/-114/- 115 A318-111/-112/-121/-122</td>
<td>ISPSS installation on A320 family. The change consists in installing and activating an In-Seat Power Supply system (ISPSS) only on FW and MB cabin areas will be filled with power outlets (36 ISP and 69 outlets).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>-MASTER DOCUMENT LIST, ref: SCA-00070-MDL-R02 dated 24/04/2017, INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, ref: SCA-00070-ICA-R01 dated 02/04/2017, AIRCRAFT FLIGHT MANUAL SUPPLEMENT, ref: SCA-00570-APMS-R08 dated 21/03/2017, or later revisions of the above listed documents approved by EASA.</td>
<td>25/04/2017</td>
<td>Active</td>
<td>10061653</td>
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<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref</td>
<td>STC ref</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>ASI INNOVATION</td>
<td>REV. 1</td>
<td>AS 332 L1</td>
<td></td>
<td>ASI ref 0284 - Installation of two Garmin GPS400XV and certification off-RNAV+ (&gt; 5° F-RNAV, RMP-APCH, and LPV-APCH)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Rotorkraft Flight Manual Supplement (RMS), dated 21 August 2015.</td>
<td>This design change cannot be installed on helicopter with the AHCAS system (commercial reference AS332L1a). The ASI modification approval n. 20/55017 must be embodied prior to installation of this design change. Prior to installation of this design change it must be determined that the interrelationship...</td>
<td>26/04/2017</td>
</tr>
<tr>
<td>21JERODESIGN LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY A321-111/-120/211-212/-213 A320-211/212/214/-215/-216 A319-111/-120/-113/-114/-115 A318-111/112</td>
<td></td>
<td>Nacelle - CFM56-5A15B Engines - Thrust Reverser - Replacement of adapting outer flange segments – Amendment of modelling</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airmus...</td>
<td>- Master Data List (MDL) 0106-00516-24growth Revisions 30September 2021 - Instructions for Continued Airworthiness, ref: 0706-00517-24growth Approval AO, dated 30 September 2021 - Electrical Wiring Inconvenien System Instructions for Continued Airworthiness, ref: 0706-00512-24growth Revis...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>14/08/2017</td>
</tr>
<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST04006NY</td>
<td>737-700, 737-900ER, -8,-9</td>
<td>Installation/Upgrade/Demod of an Inmarsat 2 Ku Antenna</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airmus...</td>
<td>- Master Data List (MDL) 0106-00516-24growth Revisions 30September 2021 - Instructions for Continued Airworthiness, ref: 0706-00517-24growth Approval AO, dated 30 September 2021 - Electrical Wiring Inconvenien System Instructions for Continued Airworthiness, ref: 0706-00512-24growth Revis...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>31/01/2017</td>
</tr>
<tr>
<td>AVIATION TECHNICAL SERVICES, INC.</td>
<td>REV. 1</td>
<td>FAA STC ST05275E</td>
<td>A319 SERIES</td>
<td>Installation of a reconfiguration of the cabin interior</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>1. Aviation Technical Services Master Data List (MDL) 7211146- MDL-001 Revision G, dated 29 March 2017, or later Federal Aviation Administration (FAA) approved review revision 2 - Instructions for Continued Airworthiness (CA) 721546-ICA-001-Revision-, dated 14 November 2016, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>26/04/2017</td>
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<tr>
<td>FALCON CREST AVIATION SUPPLY, INC.</td>
<td>REV. 1</td>
<td>FAA STC SR09034RC</td>
<td>MBB-BK117 C-1 MBB-BK117 B-2 MBB-BK117 B-1 MBB-BK117 A-4 MBB-BK117 A-3 MBB-BK117 A-1</td>
<td>Installation of a Concorde R3306E valve regulated sealed lead-acid battery</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>1. Rotorkraft Flight Manual Supplement (RMS), dated 21 August 2015.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>08/05/2017</td>
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<tr>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>Pratt &amp; Whitney Engine Services, Inc.</td>
<td>FAA STC SA1115SSC</td>
<td>B300C B300 B200GT B200C B200 300LW 300</td>
<td>Installation of Pratt &amp; Whitney flight data acquisition, storage, and transmission (FAST) system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Data List Installation Data, for King Air 200 &amp; 300 Series, document no. GN4-T-100-4, revision H, dated 14 November 2016 and/or later FAA approved revision. 2. Aircraft Flight Manual/Supplemental for King Air 200 &amp; 300 Series, document no. GNA-T-250-5, revision B, dated 27 October 2016 and/or later FAA approved...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/05/2017</td>
<td>Active</td>
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<tr>
<td>Beringer Aero</td>
<td>FAA STC 2016-75</td>
<td>LS4 L54-4 A LS4-8 LS4-84 A LS4-8-B</td>
<td>Installation of a rigid cargo barrier on the main deck in the forward cabin.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments. If indicative on FAA STC, there are requirements for environmental protection and the associated certificated noise and/ or emissions levels of the product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Master Data List Bemerdar, CL-600-2B19 (CRJ-100/200) Rigid Cargo Barrier, Report No. VA-MDL-1000, Revision A, dated 13 July 2016 or later approved FAA revision. 2. FAA accepted Bombardier CL-600-2B19 Maintenance Manual Rigid Cargo Barrier – CRJ-100/200, including instructions for Continued Airworthiness.</td>
<td>1. This cargo barrier may only be installed on Bombardier CL-600-2B19 aircraft which have a main deck cargo door and a Class B cargo compartment installed per FAA STC 20608LA and cargo handling system per FAA STC 20603LA. Prior to installation of this design change it must be determined that...</td>
<td>01/03/2024</td>
<td>Active</td>
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<tr>
<td>Ventura Aerospace, Inc.</td>
<td>FAA STC ST02670LA</td>
<td>CL-600-2B19 (RJ SERIES 100)</td>
<td>Installation of a rigid cargo barrier on the main deck in the forward cabin.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments.</td>
<td>1. Master Data List Bombardier, CL-600-2B19 (CRJ-100/200) Rigid Cargo Barrier, Report No. VA-MDL-1000, Revision A, dated 13 July 2016 or later approved FAA revision. 2. FAA accepted Bombardier CL-600-2B19 Maintenance Manual Rigid Cargo Barrier – CRJ-100/200, including instructions for Continued Airworthiness.</td>
<td>1. This cargo barrier may only be installed on Bombardier CL-600-2B19 aircraft which have a main deck cargo door and a Class B cargo compartment installed per FAA STC 20608LA and cargo handling system per FAA STC 20603LA. Prior to installation of this design change it must be determined that...</td>
<td>07/05/2018</td>
<td>Active</td>
</tr>
<tr>
<td>Jet Avionics Aviation &amp; Marine Service Center B.V.</td>
<td>FAA STC ST02600LA</td>
<td>AS350 B3</td>
<td>Installation of a rigid cargo barrier on the main deck in the forward cabin.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments.</td>
<td>1. Master Data List Bombardier, CL-600-2B19 (CRJ-100/200) Rigid Cargo Barrier, Report No. VA-MDL-1000, Revision A, dated 13 July 2016 or later approved FAA revision. 2. FAA accepted Bombardier CL-600-2B19 Maintenance Manual Rigid Cargo Barrier – CRJ-100/200, including instructions for Continued Airworthiness.</td>
<td>Only applicable to AS350 B3 S/N 34835-1X-embled.. - EASA STC 10016711 “Onboard Systems Cargo Swing” - Boggi S.r.l. SB 350 P0080-00 for cable routing openings - Airbus Helicopters SB 25.00.04 for the “Installation of 450 lb..</td>
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<td>10061698</td>
<td>AVIONIK STRAUßING ENTWICKLUNGS GmbH</td>
<td>F172P</td>
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<td>Installation of Provisions for Mission Equipment Carriage</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s): at a later amendment: CS-23, Ammd. 429, 301, 303, 306, 307, 327, 561, 601, 603, 605, 607, 609, 611, 612, 615, 1371, 1396, 1398, 1399, 1327, 1.,...</td>
<td>- Flight Manual Supplement AIR-2016-008-FMS-05-00, Rev. 00 later revisions of the above listed documents approved by EASA - Certification Program and Declaration of Conformance AIR-2016-008-CEP-03-00, Rev. 00- Project Description AIR-2016-008-PRD-04-00, Rev. 00- Instructions for Installation AIR...</td>
<td>28/04/2017</td>
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<td>10061714</td>
<td>TEXTRON AVIATION Inc.</td>
<td>REV. 2</td>
<td>SEE FAA AML E. T206H E. T208 D. T182T C. T172 T172S B. G58 A. G36</td>
<td>Installation of the G1000 Integrated Autonics System hardware and software updates, including optional Flight stream 5/10.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for the environmental protection and the associated certified noise and/ or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Garmin Master Drawing List (MDL) w/005-0620-21, Rev. 1, dated December 20, 2016. Operate in accordance with Garmin Aircraft Flight Manual Supplement assigned in the MDL and EASA AFMS ref 190-02126-06-01, Rev.1, dated March 03, 2017. Operate in accordance with Garmin Inst. Prior to installation of this design change it must be determined that the inter...</td>
<td>No.</td>
<td>15/06/2021</td>
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<tr>
<td>10061729</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>A330-243</td>
<td>A330-200 Cabin rearrangement/Cabin seating rearrangement including &quot;ACRO series 3&quot; /V passenger seats and new lavatory/L53 installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for the environmental protection and the associated certified noise and/ or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- Master Data List. Change Approval Sheet ref: CAS- A330-18-010 Rev 0. Instruction for Continued Airworthiness, ref: ICA, A330-16-010 Rev 0 or later revisions of the above listed documents approved by EASA.</td>
<td>According to the design related to this STC, This modification must be embodied in conjunction with EASA STC 02580 (Introduction of self-Illuminated Floor Path Marking System - colored guidelinePL88-900 Series) manufactured by Lufthansa Technik AG (LHT) Prior to installation of this design change it...</td>
<td>No.</td>
<td>03/05/2017</td>
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<td>10061774</td>
<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>FAA STC ST04281AT</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation for the Upgrade of the N2300 FMS computer for softwareversion N25.1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval as amended by the following paragraphs: Regulations at the latest amendment 25-0 through 25-14225.111(1) &lt;p&gt;25.153(1)(&lt;p&gt;25.1301(1) &lt;p&gt;25.1317(1) &lt;p&gt;25.1335(1) &lt;p&gt;25.1431(1) &lt;p&gt;25.152...</td>
<td>1. Chippewa Aerospace Master Drawing List (MDL) No. CAU-3461-01-0001 Rev. 1, dated 8 December 2016, or later FAA approved revision 2. Instructions for Continued Airworthiness (ICA), Chippewa Aerospace, as listed on the AML, or later FAA approved revision 3. Document No. CAU-3461-01-0002, Rev. IRA...</td>
<td>1. FAA Approved Airplane Flight Manual Supplement (AFMS) related on AML or later FAA approved revision are a required part of this STC 2. The STC is valid for those aircraft types and models approved by EASA within the FAA AML No. CAU-3461 Rev IR, dated 8 December 2016.3. Prior to installation...</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>10061791</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST00437E-D</td>
<td>LEARJET MODEL 45</td>
<td>Installation of a Universal Avionics Systems Corporation, Terrain awareness and Warning System (TAWIS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Learjet Inc. Master Document List BAS4501007, revision (1), dated 1 February 2001, FAA approved 8 February 2001 or later approved revision, 2. Instructions for Continued Airworthiness, BAS-4501007-IFCA, dated 30 January 2001, FAA approved 30 January 2001 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/05/2017</td>
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<tr>
<td>10061793</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST00437E-D</td>
<td>LEARJET MODEL 45</td>
<td>Installation of a Filter in the Cabin</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. FAA Approved Learjet Inc. Master Document List BAS4501011, Revision (6), dated 17 January 2001, or later FAA approved revision, 2. Instructions for Continued Airworthiness, FAA approved BAS-4501011-IFCA, dated 21 December 2000, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/05/2017</td>
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<td>10061793</td>
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<td>10061795</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01129W-D</td>
<td>LEARJET MODEL 45</td>
<td>Installation of a Filter in the Cabin Temperature System on Learjet Model 45.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Learjet Master Document List M100385ICT, revision (A), dated 14 November 2002, FAA approved 14 November 2002, or later approved revision, 2. Instructions for Continued Airworthiness, M100385ICT-IFCA, revision (C), dated 14 November 2002, FAA approved 14 November 2002 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10061796</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01129W-D</td>
<td>LEARJET MODEL 45</td>
<td>Installation of a Filter in the Cabin Temperature System on Learjet Model 45.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Learjet Master Document List M100448ICT, revision (C), released 8 May 2004, FAA approved 18 April 2003, FAA approved 14 November 2002, or later approved revision, 2. The following document is required for MDL M100448ICT-001 and -002: configurations of this STC: FAA approved Airplane Flight Manual Supplement, W/1533, revision (C), FAA approved.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/06/2017</td>
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<td>10061800</td>
<td>ARMSSTRONG AEROSPACE Inc.dbA ASTRONICS ARMSSTRONG AEROSPACE</td>
<td>FAA STC ST04131CH</td>
<td>A320-211/-212/-213/-216/-232</td>
<td>Installation of a Zodiak R/A/Een-Flight Entertainment System in accordance with StandardAeroMaster Data List DC104-9739-00, Revision (V) dated 6th March 2017 or later approved revisions.</td>
<td>The Certification Basis for the original product is amended to allow the following additional or alternative avionics hardware requirements: C5 25 impacted requirements at the interim stage unless otherwise specified 25, 1 25,301 (a) (c) (amendment 1), 25,307(a)(amendment 1) 25,567(a)(c), 25,57(a)(c)</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>See those of original FAA STC ST04131CH and those of the Airplane Flight Manual Supplements and other documents listed above under associated electrical documentation. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other...</td>
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<td>LEARJET Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST01496W-D</td>
<td>LEARJET MODEL 45</td>
<td>Installation of an UNS-1Ew Flight Management System on Learjet Model 45. Revision 1 updates the AMS and ICA.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Approved Master Document List M100520CT, revision (D), dated 21 December 2008; FAA approved 22 December 2008; updated FAA approved revision 2. The following document is required for M100520CT-001 configuration of this STC.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/02/2021</td>
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<td>LEARJET Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST01718A-D</td>
<td>LEARJET MODEL 45</td>
<td>Installation of a Universal Avionics System (TAWS) SCN 11.1 Upgrade, a Glass &quot;A&quot; System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Approved Master Document List BAS45010020, Revision F, dated 28 March 2005, or later FAA Approved Revision.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/05/2017</td>
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<td>TURK HAVA YOLLARI TEKNIK A.S., TURKISH TECHNIC Inc. )</td>
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<td>737-800</td>
<td>Introduction of First Row CVS Seats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Programme C0067-CEP-02.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/05/2017</td>
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<td>LEARJET Inc.</td>
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<td>FAA STC ST01734A-D</td>
<td>LEARJET MODEL 45</td>
<td>Installation of a Cockpit Speaker Mute System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Approved Master Document List BAS45010020, Revision F, dated 10 February 2007 or later FAA Approved Revision.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/05/2017</td>
<td>Active</td>
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<td>2 EXCEL DESIGN LIMITED</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>B200CT, B200T</td>
<td>Introduction of antennas and pressure hull feed through provisions and installations.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>To be installed according to instructions in the Master data list in D0119081097(1) Issue 1 dated 26th April 2017 to be maintained according to Instructions for Continued Airworthiness(2X019,019 LEFT3(1) Issue 1 ICA dated 14th December 2016. No AFM Supplement is issued with this STC or later revisions of the above.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/05/2017</td>
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<td>STC TWENTY ONE LIMITED</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>CL-800-2B16 (604 VARIANT)</td>
<td>Upgrade to Cabin Interior.</td>
<td>The Certification Basis for the original product as amended for: Repairing additional or alternative airworthiness requirements - Special Condition (s): CRI D-01 GLASS SCREENS OF DISPLAYS/MONITORS. The requirements for environmental protection and the associated certified noise and/or emissions levels...</td>
<td>STC Twenty One Ltd Flight Manual Supplement No. S21 FMS-0146 Issue 1 for later revisions of the above listed documents approved by EASA STC Twenty One Ltd Master Document List No. S21 DRL-1295 Issue 1.</td>
<td>It is the responsibility of the operator to develop procedures for developing and maintaining a list of authorized PEDs for use on the aircraft. This condition also requires the operator to list each PED (type) and maintain a list of PEDs found acceptable for use.</td>
<td>10/05/2017</td>
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<td>AERODATA AG</td>
<td>B200C</td>
<td>Enhancement of TDR94/94D by ACS-B Capability.</td>
<td>- Installation according to Service Bulletin SB-B200-TDR-2159 Rev. 2 Operations according to Airplane Flight Manual Supplement AFM-159 Rev. 0 - Maintenance accordance to Aircraft Maintenance Manual Supplement AMM-1166 Rev. 0 or later revisions of the above listed documents approved by EASA.</td>
<td>Limited to aircraft equipped with - Rockwell Collins Pro Line 21 avionics - Rockwell Collins GPS 4000S (Single) - GPS-3ARINC 429 HS Bus - TDR94 (622-9352-xxx) / TDR94D (622-9210-xxx) - xxx = 008 / 108 / 308 / 408 / 409 , Prior to installation of this design change it must be determined that it is...</td>
<td>10/05/2017</td>
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<td>FOKKER SERVICES B.V.</td>
<td>737-800</td>
<td>Boeing 737 provisions for EFB.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- CP-0234 Issue 1- CRR-0234 Issue 1- Instructions for Continued Airworthiness Supplement SE-804 Issue 1 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interfare between this design change and any other previsouly installed design change and/or repair will introduce a...</td>
<td>10/05/2017</td>
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<td>J &amp; C AEROSPACE DESIGN AND CERTIFICATION SERVICES EEU Ltd.</td>
<td>737-800</td>
<td>Change Interior Arrangement from 166 Pax (12/154Y) to 176 Pax (8J/168Y) and associated modifications.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List, Ref JCAERO/MDL/0162 Issue 1 or later revisions of the above listed documents approved by EASA.</td>
<td>- The allowed preconfiguraations are defined in the CertificationProgramme Ref JCAERO/MDL/0162 Issue 1 and are approved by PauasianAvrex Gmbh Minor Change Ref MCA-D048 or J-4 (V)-C-10 (-&gt; 0) Aero Minor Change Ref JCAERO/MOD/D/ 0175- Only applicable to Boeing 737-800 (SN 28239, 28243, and 30840). Prior to inst...</td>
<td>10/05/2017</td>
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<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 4</td>
<td>Installation of Rockwell Collins, Inc Large Display System(LDS)Revision 1 renders the 10.02 Instruction Revision 2 adds Approach Mode to the ND Format, corrects Cabin pressure controlled by Barometric Correction Settings. LDS database established moves from parking brake to the left engine fuel...</td>
<td>See EASA approved Annex</td>
<td>Master Drawing List (MDL) RCA-0415, Revision R, dated 29 October 2019 or later approved revision Table 1 - Boeing 767-300F Series Aftermarket Instructions for Continued Airworthiness RCA-0432, Revision (-), dated 29 May 2014, or later accepted revision; Airplane Flight Manual Supplement 3422-816- AFM, RL.</td>
<td>Other (1) Aerosanes modified with this STM must have the Engine Indication and Crew Alerting System (EICAS) computer Operational Program Software OPS) part number 31/5B-CDL-EOO-08 installed in the Left A Right EICAComputers (2) Intermix of General Electric CF6-80A engines with CF6-804 engines not...</td>
<td>02/09/2020</td>
<td>Active</td>
<td>10061829</td>
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<td>BLR AEROSPACE, I.L.C.</td>
<td>REV. 5</td>
<td>Approval of Installation of vortex generators and operational screwing weights.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Installation of vortex generators, manufactured in accordance with the following instructions ref AA1161, Revision B, dated November 15, 1993, and installed in accordance with Document RL-1162, Revision A, dated August 31, 1993 operation in accordance with the following instructions ref AFMS-NAV-5, dated November 18, 1993, or...</td>
<td>Serial Numbers eligible 31-2 and subsequent All (limitations) Conditions defined on the original FAA STC SA00037SE must be respected Prior to installation of this design change it must be determined that the...</td>
<td>12/05/2017</td>
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<td>10061847</td>
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<td>AERODATA AG</td>
<td>B200GT</td>
<td>Conversion for Maritime Patrol Mission Ref. 1: AFM-163 includes the content of Temporary Revision AFMTR-116-4. Correction of Drop Hatch Operation.</td>
<td>See EASA approved Annex This certification/approval involves an impact on the noise database and/or emission levels. The environmental protection requirements are amended as follows: ICAO Annex 16, Volume I, 6thedition, amended...</td>
<td>Master Drawing List MDL/M-V/893 Rev. 7Airplane Flight Manual Supplement AFM-163, Rev 5 and Maintenance Amendment AMM-1164 Rev. 2or later revisions of the above listed documents approved/accepted under the EASA system.</td>
<td>- Limited to B200GT aircraft with installed - Rockwell Collins Pro Line 21 avionics - Aerosdata STC 100806EZ Rev. 1 or later (increase of MTOW from 14000lbs) Prior to installation of this change/repair it...</td>
<td>19/02/2024</td>
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<td>10061854</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
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<td>FALCON 2000</td>
<td>Installation of Antenna for XM Weather System</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CS25.Amt 18 for the affected area as detailed in the compliancechecklist, Ref. FA03-CHK01 issue 117.</td>
<td>-Master Drawing List, Ref. FA03-RPT02 Issue 2 - MaintenanceManual Supplement, Ref. FA03-HD02 Issue 4 or later revisions of the above listed documents approved by EASA.</td>
<td>As per Maintenance Manual Supplement, Ref. FA03-HD02 Issue 4, Section 11 (Airworthiness Limitations). Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/05/2017</td>
<td>Active</td>
<td>10061854</td>
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<tr>
<td>10061857</td>
<td>FOKKER TECHNIK B.V.</td>
<td>REV. 1</td>
<td></td>
<td>A319-115</td>
<td>Installation of a GOGO satellite communication antenna</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Equivalent Safety Finding(a) CR-112 Vibration / buffeting compliance criteria for large transportaircraft installation and/or requirements for environmental protection and th...</td>
<td>-J319-AEI-MS-002, Revision 2 - A319-GOGO-ICA-0002, Issue 1or later revisions of the above listed documents approved by EASA. EBA319-0046, Engineering Bulletin, First release - CP-0342, Certification Plan, Issue 2 - ECR-0342, Compliance Record Report, Issue 1.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/12/2019</td>
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<td>10061874</td>
<td>TENENCIA Ltd.</td>
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<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>Electronic Flight Bag (EFB)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>TEN 817-47, Aeronautical Flight Manual Supplement, Issue 1or later revisions of the above listed documents approved by EASA. TEN 817, Master Data List, Issue 1.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/05/2017</td>
<td>Active</td>
<td>10061874</td>
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<td>10061878</td>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY NTER USA, INC</td>
<td>REV. 2</td>
<td></td>
<td>EMB-135BJ</td>
<td>Installation of a Forward Lavatory with an Externally Served WasteSystem. This STC was originally issued as a validation of Brazilian CTS (STC) No. 2010S08-11 (16 August 2010), and transferred to a SubstanceList by Embraer Design Approval Holdend 22 July 2019.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>- Embraer Installation Manual, EIC/WWE/001, Revision B, dated22 January 2016, Master Document List No. EIC/WWE/001, Rev. 5, dated 9 October 2018., Instructions for Continued Airworthiness (ICA), Document No. ICA-008 Rev. A, dated 9 October 2018 for later revisions of the above listed document(s) approved...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/11/2019</td>
<td>Active</td>
<td>10061878</td>
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<td>10061893</td>
<td>AVON INDUSTRIES, Inc.</td>
<td>REV. 1</td>
<td></td>
<td>FAA STC ST116-010NM</td>
<td>Installation of Removable External Baggage Compartment.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Avon Maintenance Manual, dated 20 October 1987, or Instructions for Continued Airworthiness, No. 03607200, Rev. IR- (+) or dated 15 February 2017, and later FAA accepted revision.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. The STC is valid for those aircraft types...</td>
<td>19/05/2017</td>
<td>Active</td>
<td>10061893</td>
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<tr>
<td>10061899</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td></td>
<td>SAAB SF340A, SAAB 340B</td>
<td>Upgrade of Universal (UNS) Flight Management System, including LPI/monitor &amp; Rockwell Collins Transponder for Mode S EL/ELHS &amp; ADS-B OUT. This STC consists through various optional upgrade in modifying the existing single Universal UNS-11 FMS system to a Universal UNS-11e FMS (option A), eventually used...</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CS25.Amt 2...</td>
<td>- SAD Engineering Order, ref. 204950-EO Issue 1 for FMS, ref. 204950-AM A Issue 1 for LPV Monitor, ref. 204950-AM B Issue 3 for XPDR, ref. 204950-AM C Issue 15AD Instruction for Continued Airworthiness, for FAA, ref. 204950-CA A Issue 1...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/11/2017</td>
<td>Active</td>
<td>10061899</td>
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<td>Rev.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>Model(s)</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td></td>
<td></td>
<td>B300</td>
<td>Installation, Operation and Maintenance of Air Ambulance Technology AE</td>
<td>Equipment Beech 300 EMS installation PN B300-25-20-103.000-600 in accordance with MEL</td>
<td>Flight Manual Supplement FMS-Beech300-EMS-000 Air Ambulance/Technology PN B300-25-20-103.000-600 Initial Issue, dated April 24, 2017, or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/05/2017</td>
<td>Active</td>
<td>10061912</td>
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<tr>
<td>AVDYNE CORPORATION</td>
<td></td>
<td>FAA STC SA0362BO</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of Globalstar Antenna and Cradle provisions to support a Globalstar GSP-1700 in non-composite airplanes.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>1. Avdyne Corporation Master Document List (MDL)-&lt; &gt; Document Number FDC-104, Revision 09, dated 9 January 2017, or later FAA approved revision 2. Avdyne Corporation Aircraft Flight Manual Supplement (AFMS) DocumentNumber 600-0027-000, Revision 05, FAA approved on 5 April 2017, or later FAA approved revision.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/06/2017</td>
<td>Active</td>
<td>10061913</td>
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<tr>
<td>SBC AMERICA LLC</td>
<td>REV. 1</td>
<td>FAA STC SA0787BE</td>
<td>737-900/-900ER, 737-700/-700/-800/-900/-1000/-2000/-300/-400/-500</td>
<td>Installation of a Sliding Carpet Cargo Loading System in the cargo compartment.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>1. Telair International Drawing List/Report List STC/03-OC, Revision 29, dated 17 December 2018, or later FAA approved revision 2. Chapter 19, Airworthiness Limitations of Technical International Sliding Carpet Manuals, Revision 2, dated 15 December 2018, or later FAA approved revision.</td>
<td>1. This approval does not include FAA approved revision 3.</td>
<td>11/03/2019</td>
<td>Active</td>
<td>10061919</td>
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<tr>
<td>HONEYWELL INTERNATIONAL (BENDIX/KING)</td>
<td>FAA STC SA11718C</td>
<td>SEE AML</td>
<td>Installation of Bendix/King by Honeywell KSN 770 and KSN 765/WASS-400 Approach capable nav/comm systems</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>1. Installation Manual 006-10716-0000, Revision 5, dated 11 July 2014, or later FAA approved revision. 2. Aircraft Flight Manual Supplement, 000-0024-001, Revision F, dated 25 July 2014, or later FAA approved revision 3. ICA 90-0002-010, Revision B dated 7 January 2014 or later FAA Approved.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and model...</td>
<td>17/05/2017</td>
<td>Active</td>
<td>10061922</td>
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<tr>
<td>AVEO ENGINEERING GROUP S.R.O.</td>
<td>REV. 1</td>
<td></td>
<td>737-900, 737-900ER, 737-700/-700/-800/-900/-2000/-300/-400/-500</td>
<td>Design Change for Boeing 737 Light replacement. Replacement of existing Navigation, Lo-to-Wing illumination, Trend or Turn lights with AVEO Engineering LED lights. Revision 1 is the addition of replacement of light bulbs for strobe lights, the anti-collision lights and the landing light and replaces the...</td>
<td>The Certification Basis for the original product as amended, this supplements the existing additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS 25 AMDT 18: 25.301-20.305, 25.609, 25.773, 25.1301, 25.1303, 25.1315, 25.385, 25.1387, 25.1389, 25.1391, 25.1393, 25.1395, 25.1397...</td>
<td>The change is defined with reference of the Master Document List(AVL) MOD-002-404, MDL-&lt; &gt; or later revisions of the above listed documents approved/accepted under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/11/2018</td>
<td>Active</td>
<td>10061932</td>
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<tr>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td></td>
<td>A340-211</td>
<td>Cabin Refurbishment.</td>
<td>The Certification Basis for the original product as amended, this supplements the existing additional or alternative airworthiness requirements. Deviation(s): Per CRI A601. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/approval.</td>
<td>Master Data List MDL-FC01-25800, Rev. A dated 16/05/2017, or later revisions of the above listed documents approved by EASA.</td>
<td>Limited to MSN 025. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/06/2017</td>
<td>Active</td>
<td>10061934</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
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<th>Description</th>
<th>Certification basis</th>
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<th>Limitations/Conditions</th>
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<tbody>
<tr>
<td>ARMSTRONG AEROSPACE Inc.d/b/a ASTRONOMG AEROSPACE</td>
<td>737-400-900-900ER/8-0</td>
<td>Installation and activation of an Astronics seat/flight system and/or FlightDeck USB UltraLite Power System (Provisions for Class 2 EFB Systems).Rev. 1 adds the models 737-8 and 737-9 to EASA approved installation eligibility and changes the address. Rev. 2 corrects some typos.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements for 737 Classic and NO models C9.26 impacted requirements at initial and unless otherwise specified C9.26(a): 25.301(a), 25.303(a) and (b), 25.307(a).</td>
<td>Astronics Armstrong Aerospace Document No. per Section 6.1, of Master/Data List DC104-2017-00, Revision W, dated 22 July 2019, and the Flight Manual Supplement Document No. DC104-2017-00, Revision W, dated 22 July 2019, or later FAA approved revisions.</td>
<td>This STC installation approval does not include the Electronic FlightBag computers. The operational approval of the electronic FlightBag computers should follow the guidance and requirements laid down in the applicable operational rules. Prior to installation of this design change it must be determined...</td>
<td>09/05/2022</td>
<td>Active</td>
<td>10161932</td>
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<tr>
<td>MONOGRAM SYSTEMS</td>
<td>FAA STC</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of the Toilet Assembly.</td>
<td>1. Prior to installation of this design change it must be determined whether the installation eligibility and changes to the following additional or alternative airworthiness requirements (CA), as noted on the FAA Approved Model List ST02624LA,</td>
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<tr>
<td>CLAY LACY AVIGATION, INC.</td>
<td>FAA STC ST0423CH</td>
<td>Installation of a Universal Avionics UNS-“E” flight management system with Link 801 datalink system and CVR-120A cockpit voice recorders meet airworthiness requirements for FANS 1/A+ and optional Automatic Dependent Surveillance Broadcast (ADS-B) Out.</td>
<td>The certification basis for the original product as amended the following additional or alternative airworthiness requirements: C9.26 impacted requirements at initial and unless otherwise specified C9.26(a): 25.301(a), 25.303(a) and (b), 25.307(a).</td>
<td>Clay Lacy Aviation Master Data List E13006-2320-L001, Revision E, dated 28 February 2017, or later FAA approved revision 2.</td>
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<tr>
<td>BLACKHAWK AEROSPACE TECHNOLOGIES INC</td>
<td>FAA STC ST01823W</td>
<td>Installation of Garmin GTS Traffic Sensor with Garmin GTX3000 Mode S Transponders with ADS-B Out.</td>
<td>The certification basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Columbia Avionics &amp; Aircraft Services Flight Manual Supplement, Document No. 30142347, Revision R, dated 16 September 2016, or Document No. 30142269, Revision R, dated 16 September 2016 and, 2. Columbia Avionics &amp; Aircraft Services Master Data List, Document No. 30142251, Revision R, dated 19 September 2016.</td>
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<tr>
<td>AVENGER AEROSPACE SOLUTIONS, INC.</td>
<td>HS-125 SERIES 400B/400A</td>
<td>Installation of Hawker Nose Landing Gear Plunger Tube PIN NP-UN49-1 (replacement for OME PIN 25UN49)</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements. Following paragraph(s) at a later amendment: JAR 25 Change 13 forth the following paragraphs: 25.301, 25.303, 25.307(a) and (b), 25.307(d) and (e), 25.301(b) and (c), 25.307(a)(c) and (d).</td>
<td>Part in accordance to drawing number NP-UN49 Rev. B, limitations as per document Airworthiness Limitation Section: AV/6-ALS-18-007, Rev. NC, dated 28/04/2016 or later revisions of the list approved by EASA in accordance with EASA 2004/04/CF (or subsequent revisions thereof)</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>TURBINE CONVERSIONS, LTD</td>
<td>REV. 2</td>
<td>FAA SA04034CH</td>
<td>B. 206H, T206H</td>
<td>Installation of Pratt &amp; Whitney Canada PT6A engines (ref. to Limitation/Section) and Honeywell HT-837R-3 propeller with T10173-8 blades.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>- Turbine Conversions, Ltd. Master Drawing List Report 200601, Rev. E, dated 14 Jan 2015.</td>
<td>10/30/2017</td>
<td>Active</td>
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<td>CLAY LACY AVIATION, INC.</td>
<td>FAA STC</td>
<td>ST04514CH</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Installation of dual Universal Avionics UXi-Fire flight management systems with UXi-L6-UL-801 datalink system and CVR-120A or CVR-120R cockpit voice recorder to meet environmental requirements for FANS 1A + NEW PAGE.</td>
<td>The certification basis for the original product as amended by the following additional or alternative airworthiness requirements: 14 CFR Part 25.25.86(a)(4) [Amend 25-11]; 25.1209(c), (d), (e) [Amend 25-11]; 25.1457(a), (b), (c), (d), (e), (f), (g), (i) [Amend 25-11]; 25.1529(a), (b), (c), (f), (g), (h), (i), (j), (k), (l), (m), (n), (o), (p), (q), (r), (s), (t), (u), (v), (w), (x), (y), (z) [Amend 25-11].</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. The modified aircraft must be maintained in accordance with the instructions for continued airworthiness.</td>
<td>29/05/2017</td>
<td>Active</td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>FAA STC</td>
<td>ST04362AT</td>
<td>A321-231/231-232</td>
<td>Installation of the Rockwell Collins Automatic Dependent Surveillance Broadcast (ADS-B) Out System on the Airbus A319/320/321 Series Aircraft.</td>
<td>The certification basis for the original product as amended by the following additional or alternative airworthiness requirements: 14 CFR Part 25.25.23(a), 25.26.95(a)(4), 25.26.95(b), 25.26.95(c), 25.26.95(d), 25.26.95(e), 25.26.95(f), 25.26.95(g), 25.26.95(h), 25.26.95(i), 25.26.95(j), 25.26.95(k), 25.26.95(l), 25.26.95(m), 25.26.95(n), 25.26.95(o), 25.26.95(p), 25.26.95(q), 25.26.95(r), 25.26.95(s), 25.26.95(t), 25.26.95(u), 25.26.95(v), 25.26.95(w), 25.26.95(x), 25.26.95(y), 25.26.95(z).</td>
<td>1. Master Drawing List (MDL) 2-177034-001, Rev. A, dated 31 March 2017, or any other FAA approved revision.2. FAA Approved Airplane Flight Manual Supplement for the Rockwell Collins Automatic Dependent Surveillance Broadcast (ADS-B) Out System.2-177034-001, Rev. IR, dated 2 May 2017, or any other FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/05/2017</td>
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<tr>
<td>AVIATION SUPPORT GmbH</td>
<td>SA 365 N</td>
<td>767-300</td>
<td>767 Cabin Attendant Crew Rest Seat</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Flight Manual Supplement, FMS-ASE-91616, original release or later revisions of the above listed documents approved by EASA - Liste der gültigen Dokumente, MLD-ASE-91616, original release, dated 15/05/2017. Instructions for Continued Airworthiness, ICA-ASE-91616, original release, dated 15/05/2017.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/05/2017</td>
<td>Active</td>
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<td>SW6 CERTIFICATION SERVICES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>767-300</td>
<td>767 Cabin Attendant Crew Rest Seat</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>SW6 Certification Services Ltd Master Certification Scheme No.3513-MCS-01 Issue 1</td>
<td>None</td>
<td>22/05/2017</td>
<td>Active</td>
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<td>Rev. level</td>
<td>Foreign/NA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10061969</td>
<td>JET AVIATION AG, BASEL</td>
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<td>MYSTERE FALCON 900 (MF900)</td>
<td>VIP &amp; Medevac cabin reconfiguration</td>
<td>The Certification Basis for the original product as amended (following additional or alternative airworthiness requirements: Special Condition (s): U/D-17 - No Smoking Signs and Lavatory Ashtrays D-18 - Width of Aisle) The requirements for environmental protection and the associated certified noise and emissions levels of the original product remained unchanged and remained applicable to this certificate/ approval.</td>
<td>- Master Data List JBSL-M-00727MDL-01 Issue 2 dated 18.05.2017.AFM Supplement No. JBSL-M-00727AFM-01 Issue 2 dated 12.09.2017. Instructins For Continued Airworthiness JBSL-M-00727CA-01 Issue 2 dated 15.06.2017 or later revisions of the above listed documents approved by EASA.</td>
<td>Effective AV189 SN 90007 and upwards (except SN 49004&lt;=&gt;70192) and SN 900001 and upwards (SA02452SE).</td>
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<td>10061970</td>
<td>AEROLITE AG</td>
<td>REV. 1</td>
<td>AIW189</td>
<td></td>
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<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remained unchanged and remained applicable to this certificate/ approval.</td>
<td>Certification Program CPR-21.006 revision BF3 Flight Manual Supplement FMES-189-0213100-501 revision A Master Document List MDL-15-022 revision Ac or later revisions of the above listed document(s) approved/accepted/published by the EASA system.</td>
<td>Effectively AV189 SN 90007 and upwards (except SN 49004&lt;=&gt;70192) and SN 900001 and upwards (SA02452SE).</td>
</tr>
<tr>
<td>10061971</td>
<td>AVENGER AEROSPACE SOLUTIONS, INC.</td>
<td>REV. 1</td>
<td>PMA N. PQ50318HN</td>
<td>HAWKER 900XP HAWKER 800XP HAWKER 800 HAWKER 750 BAE.125 SERIES 800B BAE.125 SERIES 800A</td>
<td>Hawker Main Landing Gear Plunger Tube PIN NP-8UM-5 / 6 / 7 / 8 (replacement for OEM PIN 25-8UM5 and 25-8UM6)</td>
<td>The Certification Basis for the original product as amended (following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: JAR 25 Change 13 for the following paragraphs: 25.301, 25.303, 25.305(a)(2)(b), 25.307(a) and (b), 25.317(a)(3) and (c)(2).</td>
<td>Part in accordance to drawing number NP-8UN Rev A Limitations as per document Airworthiness Limitation Section: AVS-ALS-16-008 Rev. A, dated 11/03/2016 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF for subsequent revisions of this decal.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10061972</td>
<td>AEROCEUT, INC.</td>
<td>FAA STC SA02452SE</td>
<td>KOENIG 100</td>
<td>Installation of Aerocet Model 6650 Amphibious Twin Floats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remained unchanged and remained applicable to this certificate/ approval.</td>
<td>1. Master Drawing Document M-00727 (MDL) Document 66-1200, Revision J dated 31 August 2015, or later Federal Aviation Administration (FAA) approved revision. 2. Instructins for Continued Airworthiness (ICA) Document A-10008, Revision 0, dated 11 February 2015, or later FAA-accepted revision. 3. Airplane Flight Manual.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/05/2017</td>
</tr>
<tr>
<td>10061979</td>
<td>TRUATLANTIC MFG., LLC</td>
<td>FAA STC SA0214AT-D</td>
<td>B) (G36 A) 635, 655, V35A, V35B A) 96, A36, E33A, E33C, F33A A) 25-D33A</td>
<td>Installation of a Stainless Steel Exhaust System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remained unchanged and remained applicable to this certificate/ approval.</td>
<td>- Atlantic Aero, Inc. Master Drawing List 1595001, Rev. F dated March 2011 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decal) and/or the Technical Implementation Procedures of ELUUSA Bilateral.</td>
<td>None</td>
<td>23/05/2017</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>SBO AMERICA LLC</td>
<td>REV. 2</td>
<td>FAA STC</td>
<td>737-80/1 SERIES 737-600-700-800-900 SERIES</td>
<td>Installation of electrical wiring provisions for cargo loading system in forward and all lower cargo compartments.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. There are requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>1. Master Drawing List B737-1000, Revision H, dated 31 March 2019; or later FAA approved revision. 2. Engineering Change Order (ECO) 737-201, Revision G, dated 10 April 2001; or later FAA approved revision. 3. For Boeing Model 737-600-700-800-900 cargo compartments and Service Bulletin 737-25-39/100000.</td>
<td>15/11/2018</td>
<td>Active</td>
</tr>
<tr>
<td>PAUSTIAN AIRTTEX GmbH</td>
<td>LBA TA0712</td>
<td>737-800</td>
<td></td>
<td>Cabin Configuration Change from 188YC to 188YC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List Ref. B737/38-25-04-02 Rev. 03 dated 2012-01-08 or later revisions of the above listed documents approved by EASA.</td>
<td>19/06/2017</td>
<td>Active</td>
</tr>
<tr>
<td>ROEDER PRAEDISON GmbH</td>
<td>M20, M20A, M20B, M20C, M20D, M20E, M20F, M20G, M20J, M20K</td>
<td>Installation of Bendix King Modo S Tranducer KT 73</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EASA STC 2004-12063 from Nayak Aircraft Service GmbH based on LBA EMZ SA 1421 issue 1, dated 10th December 2004.</td>
<td>EASA STC 2004-12063 from Nayak Aircraft Service GmbH based on LBA EMZ SA 1421 issue 1, dated 10th December 2004- AFM Supp. Doc. M20- NY-0706-08. Rev. for later revisions of the above listed documents approved by EASA.</td>
<td>As listed in the AFM. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/10/2017</td>
<td>Active</td>
</tr>
<tr>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>F28 MARK 0100</td>
<td>Introduction of Passenger Briefing SystemRev. 1. Transfer of STC</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: T-500-87 (FA 21 Issue 4). The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>ATL Engineering Order ATL/LPS-0004 Issue 1 or later revisions of the above listed document(s) approved/accredited under EASA system.</td>
<td>ATL Engineering Order ATL/LPS-0004 Issue 1 or later revisions of the above listed document(s) approved/accredited under EASA system.</td>
<td>1. Subject to compliance with the provision of Part 21 Subpart E, the STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this certificate is requested under Part 21A.118, the certificate will be released. This STC is approved only for the product.</td>
<td>10/11/2021</td>
<td>Active</td>
</tr>
<tr>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>F28 MARK 0100</td>
<td>Interior Modification for 60 seat VIP layoutRev. 1. Transfer of STC</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: T-100-87 (FA 21 Issue 4). The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged.</td>
<td>FAC Inc Modification Report 196707100 initial issue or later revisions of the above listed document(s) approved/accredited under the EASA system.</td>
<td>FAC Inc Modification Report 196707100 initial issue or later revisions of the above listed document(s) approved/accredited under the EASA system.</td>
<td>1. Subject to compliance with the provision of Part 21 Subpart E, the STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this certificate is requested under Part 21A.118, the certificate will be released. This STC is approved only for the product.</td>
<td>10/11/2021</td>
<td>Active</td>
</tr>
<tr>
<td>MT-PROPELLER ENTWICKLING GmbH</td>
<td>LBA SA1369</td>
<td>PA-60-700P (AEROSTAR 700P)</td>
<td>Extension of LBA/EASA STC SA1369 and SA1403.Aerostar PA-60-700P; Installation of MTV-14-B-CCCFs, 188-30d propellers optional installation of MTV-14-B-CCCFs, 188-30d propellers.Aerostar PA-60-600; PA-60-601; PA-60-601P; PA-60-602P all modified with FAA STC SA1659NM Installation of MTV-14-B-CCF188-30d.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-38, Amendment 4PAR 36, Amendment 36-303 this certificate/approval involves a change to the requirements for environmental...</td>
<td>According to MT-Propeller Master Documentation List No. E-2998, Rev. (Aerostar PA-60-700P with MTV-14-B-CCCFs, 188-30d) propeller to be operated in accordance with MT-Propeller AFM-5 Documents: E-996, Rev. 0.Aerostar PA-60-600; PA-60-601; PA-60-601P; PA-60-602P all modified with FAA-STC...</td>
<td>According AFTM-Spreen/pellte modified with STC SE1659NM for PA-60-600, PA-60-601P, PA-60-602P. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. 3. This STC is approved for the product.</td>
<td>18/07/2017</td>
<td>Active</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10062006</td>
<td>SWS</td>
<td>1</td>
<td></td>
<td>737-800</td>
<td>CLASS 2 FLIGHT ATTENDANT SEAT INSTALLATION ON NEOs 737-800.</td>
<td>SWS Certification Services Ltd Master Configuration Scheme No. 3154-MS-01 Issue 1.</td>
<td>- The cabin attendant seat rest configuration is not to be used during takeoff, take-off and landing phases of a flight and may only be used on the aircraft in a steady cruise condition. As the leg rest of the cabin attendant seat is fully deployed position fits into the cart space on each side of the...</td>
<td>24/02/2017</td>
</tr>
<tr>
<td>10062018</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>REV. 2</td>
<td></td>
<td>777-300, 777-300ER 777-200, 777-200LR</td>
<td>Wave Ka Band Antenna InstallationRevision 1 adds 777-300 and 777-300ER models and provides allowing compliance to ESF CRI B-101. Revision 2 adds the configuration 3 on B777-200LR aircraft types.</td>
<td>The Certification Basis (CB) for the original product as amended the following additional or alternative airworthiness requirements: Equivalent Safety Findings (ESF) CRI B-101 Vibration / Buffeting compliance criteria for large external antenna installations the following paragraph(s) at a later...</td>
<td>Initial Issue – Configuration 1 (777-200 - 200LR). Master Data List MSL - B777-5300 Revision D, dated 31st of October 2018 - Airplane Flight Manual Supplement AFMS-777-5300 Revision C, dated 27th of April 2017 - Instructions for Continued Airworthiness - ICA - B777-5300 Revision B, dated 30th of October 26...</td>
<td>15/02/2022</td>
</tr>
<tr>
<td>10062025</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>35 B3</td>
<td>650h EFIS SYSTEM INSTALLATION</td>
<td>AS 350 B3</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraph(s) at a later...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Findings (ESF) - CRI B-101 Vibration / Buffeting compliance criteria for large external antenna installations the following paragraph(s) at a later...</td>
<td>AE17019RFSM, Rotorcraft Flight Manual Supplement, Rev 0, dated 22 May 2017 AE17019OSDS, Operational Suitable Data Supplement, Rev 0, dated 04 May 2017 or later revisions of the above listed documents approved by EASA.AE17019CA, Instruction Manual Continued Airworthiness, Rev 0, 07 April 2017</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10062027</td>
<td>AIR ALLIANCE GmbH</td>
<td>35, 35A</td>
<td>Installation of Sandel SN4500 EFLS</td>
<td>35, 35A</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Aircraft Flight Manual, doc. no. 2015-012, LJS, 0506, revision 02, dated 12 May 2017 - Instructions for Continued Airworthiness, doc. no. 2015-012, LJS, 0507, revision 01, dated 15 May 2017 - Master Document List, doc. no. 2015-012, LJS, Form 2, revision 00, dated 22 May 2017 or later revisions of the above...</td>
<td></td>
</tr>
<tr>
<td>10062034</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td></td>
<td>DORNIER 228-212</td>
<td>High North Latitude Operation Modification</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Modification in accordance with Scandinavian Avionics Design Engineering Order 2106617-ESO Issue 2, dated 07.09.2017. Operate in accordance with Scandinavian Avionics Design Flight Manual Supplement 2106617-AM Issue 3, dated 15.08.2017.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>KELOWNA FLIGHTCRAFT LTD.</td>
<td>REV. 2</td>
<td>TCDS SA16-86</td>
<td>737-600, 737-400, 737-300, 737-405, 737-500, 737-300</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements.- The following paragraph(s) at a later amendment: CS-ACNS, original issue: CS-ACNS.D.ADBS.001, CS-ACNS.D.ADBS.010, CS-ACNS.D.ADBS.020, CS-ACNS.D.ADBS.025; CS-AC...</td>
<td>10062053; 10062054; 10062049; 10062042; 10062041</td>
<td>FAA approved installation of dual GPS systems in accordance with Identification pr...</td>
<td>EASA approved Installation of dual GPS systems in accordance with Identification pr...</td>
<td>13/06/2022</td>
</tr>
<tr>
<td>QPA S.R.L.</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Master Document List QPA-EVETOR-32-01/MDL, Ed. 1, Rev. 0, dated 2021-04-04; Check List QPA-EVETOR-32-01/CL, Ed. 1, Rev. 0, dated 2021-04-04; Certification of Change QPA-EVETOR-32-01/CD, Ed. 1, Rev. 0, dated 2021-04-04</td>
<td>EASA approved Installation of dual GPS systems in accordance with Identification pr...</td>
<td>FAA approved Installation of dual GPS systems in accordance with Identification pr...</td>
<td>26/01/2022</td>
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<tr>
<td>LANITZ-PRENA FOLEN FACTORY GmbH</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Master Document List ORATEX covering &amp; coating system, Doc. No. ADXC-11-01-01/00;...</td>
<td>EASA approved Installation of dual GPS systems in accordance with Identification pr...</td>
<td>FAA approved Installation of dual GPS systems in accordance with Identification pr...</td>
<td>16/09/2022</td>
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<tr>
<td>AIRFORMS, LLC</td>
<td>REV. 1</td>
<td>FAA STC SA12493A0K</td>
<td>208B 208</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Master Document List ORATEX covering &amp; coating system, Doc. No. ADXC-11-01-01/00;...</td>
<td>EASA approved Installation of dual GPS systems in accordance with Identification pr...</td>
<td>FAA approved Installation of dual GPS systems in accordance with Identification pr...</td>
<td>02/11/2022</td>
</tr>
<tr>
<td>AIRFORMS, LLC</td>
<td>REV. 1</td>
<td>FAA STC SA12493A0K</td>
<td>208B 208</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Master Document List ORATEX covering &amp; coating system, Doc. No. ADXC-11-01-01/00;...</td>
<td>EASA approved Installation of dual GPS systems in accordance with Identification pr...</td>
<td>FAA approved Installation of dual GPS systems in accordance with Identification pr...</td>
<td>02/11/2022</td>
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<td>STC Holder</td>
<td>STC Number</td>
<td>STC Holder</td>
<td>STC Number</td>
<td>Serial Numbers</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EXTIX ENGINEERED PRODUCTS INC</td>
<td>PMA No 9 P02203NM SUPPL 2</td>
<td>500N, 600N, 369D, 369E, 369F, 369FF</td>
<td>STC of EXTIX PMA E609625 (Transmission Interconnect Shaft-CriticalComponent PMA Replacement Part)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>Ext Aeronautical Report EE-16-5251.01 Rev IR of 25 Jan 2016;Continued Operation Safety (COS) System dated 21 October 2015.</td>
<td>None</td>
<td>12/06/2017</td>
</tr>
<tr>
<td>LIEBELLLE MONDE INC.</td>
<td>REV. 2</td>
<td>TCCA SA15-95</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Manufacture and installation of interior placards, signs and markings.Manufacture and installation of interior placards, signs and markings.Rev. 1: Updates to the certification basis.Rev. 2: Correction of Technical Document reference.</td>
<td>The Type and OSD Certification Bases for the original product are amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 part 11: 25.1, 25.601, 25.603, 25.605, 25.853, 25.855(f), 25.1301, 25.1501, 25.1520: The requirement(s): Master Drawing List MDL-PA-15-006 rev. MLBellte Instructions for Continued Airworthiness No. ICA-PA-15-006(Revised B) or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU Canada Bilateral Agreement.</td>
<td>Installation and servicing instructions as per ICAO document.Prior to installation of this change/repair it must be determined thatthe interrelationship between this design change/repair and anyother previsouly installed design change and/or repair willintroduce no adverseeffect upon the airworthiness of the pro...</td>
<td>06/02/2024</td>
<td>Active</td>
</tr>
<tr>
<td>AVIONITEC Ltd.</td>
<td>500N</td>
<td>Installation Garmin Navigator Removed: Garmin XA50AH Audio Panel; Bendix King KY 196A COM; Bendix King/LCN 90B GPS; Bendix King KK 165 NAV/ COM; Bendix King KEA 405 Radar/Attimeter System and Bendix King KN 62A ONE Installed; Garmin GMA35c Audio Panel; Garmin GTN650 (GPS/COM/MNAV);Garmin GTN750 (GPS/COM...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Garmin Flight Manual Supplement Dual GTN Installation: Doc. No. 7015-011, dated 04.04.2017; - Rotorcraft Flight Manual Supplement Traffic Advisory System: Doc. No. 7015-012, dated 04.04.2017 or later revisions of the above listed documents approved by EASA - Master Document List: Doc. No. 7015-01...</td>
<td>- S-N LN076 (HB-ZRM): GNSS Systems (GTN750/GTN650)</td>
<td>01/06/2017</td>
<td>Active</td>
<td>10202008</td>
</tr>
<tr>
<td>ADVENT AIRCRAFT SYSTEMS, INC</td>
<td>REV. 3</td>
<td>FAA STC SAU03175CH</td>
<td>SEE FAA AML</td>
<td>Installation of an anti-skid brake system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation instructions specified on FAA Approved Model List (AML) No.SAU03175CH</td>
<td>22/02/2018</td>
<td>Active</td>
</tr>
<tr>
<td>2 EXCEL DESIGN LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>PA-31</td>
<td>Transfer of EASA STC No. 10662065 for Activation of Wi-Fi for SATCOMSystem including Swift Broadband from STC Twenty One Ltd to 2ExcelDesign Ltd.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>STC Twenty One Ltd Flight Manual Supplement S21-250-030 Issue 2 dated 8December 2016 for later revisions of the above listed documents approved by EASA,STC Twenty One Ltd Master Document List S21-250-030 Issue 3</td>
<td>Applicable to aircraft serial number 31-740084 only. Prior to installation of this design change it must be determined thatthe interrelationship between this design change/repair and any otherpreviously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the pro...</td>
<td>27/04/2018</td>
</tr>
<tr>
<td>SPECIALIST AVIATION SERVICESLIMITE D</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>AW169</td>
<td>AW169 EMS Interiors (SAS/MOD/833).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement SMS/833-043 Issue A for later revisions of the above listed documents approved by EASA Declaration of Compliance for SAS/MOD/833 Issue B, dated 01/06/2017 Maintenance Program Supplement M/833-040 Issue A, dated 23/05/2017.</td>
<td>This installation must be installed in conjunction with LeonardoHelicopters EMS Machined Floor (UR) Kit PN 6F9332F00211.Helicopter operation requiring ditching capability is not allowed.NVIS operation requires separate design approval Prior to installation of this design change it must be determine...</td>
<td>02/06/2017</td>
<td>Active</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10062104</td>
<td>AVIO TECHNOLOGIES SRL</td>
<td></td>
<td>FA150K, FA150L, FA150M, F160H, F152, FA152, F150H, F150L</td>
<td>Union Aviation Hand Control Installation (Assembly Drawing List 101 model C)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>AFM Supplement AT-STC-001-AFM, Issue 01r1 (&lt;--&gt;for later revisions of the above listed documents approved by EASA. Declaration of Similarity dated 17.05.2017) Compliance Checklist AT-STC-001-C523, Issue 01r1 - Maintenance Manual Supplement AT-STC-001-ICA, Issue 01 - STC Analytical Evaluation AT-91...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/06/2017</td>
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<tr>
<td>10062124</td>
<td>JET AVIATION AG</td>
<td>REV. 1</td>
<td>737-700 IGW</td>
<td>VVIP Cabin Interior Completion/Revision 1: Transfer STC 10062124 Revision 0 from Jet Aviation AG, Base/Completions to Jet Aviation AG</td>
<td>See CRI A-01 EASA Supplemental Type Certificate Basis on B737-700 VVIP Interior Installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Jornac Flight Supplemental Supplement; JBSC-M-00048FA-AT, Issue 02 - Certification Compliance Sheet; JBSC-M-00048CC-01, Issue 05 - Master Data List; JBSC-M-00048MDL-01, Issue 06 - Instructions for Continued Airworthiness; JBSC-M-00048ICA-01, Issue 04 - Maintenance Planning Document Supplement; MS0004-MFGS.</td>
<td>Limited to S/N 39000 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/12/2017</td>
</tr>
<tr>
<td>10062139</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>REV. 2</td>
<td>FAA ST03360CH</td>
<td>737-900, 737-900ER 737-800, 737-700, 737-800</td>
<td>Provisions for installation of Rosemount Aerospace Electronic Flight bag(EFB) systems Rev: 1: added configurations P, Rand T Rev: 2 amends configurations R and T to add capability for LRUs dataloading functionality.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1) FAA Approved Flight Manual Supplement as identified in RosemountAerospace Master Data List, Doc. No. D06414246, Revision AT, dated December 1, 2022, or later FAA Approved revision. 2) For Rosemount Aerospace Master Data List, Doc. No. D06414246, Configuration C (Provisions for Cessna 2 EFB). FAA A...</td>
<td>14/03/2023</td>
</tr>
<tr>
<td>10062142</td>
<td>IAB FL TECHNICS</td>
<td></td>
<td>737-500</td>
<td>Passenger Cabin seat reconfiguration and cabin refurbishment on Boeing737-500 aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FL TECHNICS MASTER DATA LIST No. MDL-238-01 Issue A dated 08 June 2017</td>
<td>Restricted to ASN 28923</td>
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<tr>
<td>10062144</td>
<td>TURK HAVA YOLLARI TEKNIK A.S (TURKISH TECHNICS Inc.)</td>
<td>REV. 1</td>
<td></td>
<td>Air Transport Stretcher Rev: 1: Extension of applicability of the AML STC 10062144 to include the B737-80-9 MAX and the A321NEO models</td>
<td>The Type and OSD Certification Bases for the original product are amended by the following: 1) FAA Approved Flight Manual Supplement as identified in RosemountAerospace Master Data List, Doc. No. D06414246, Revision AT, dated December 1, 2022, or later FAA Approved revision. 2) For Rosemount Aerospace Master Data List, Doc. No. D06414246, Configuration C (Provisions for Cessna 2 EFB). FAA A...</td>
<td>Certification Programme C0056-CEP Rev. 3 dated 17.02.2023 Component User Manual C0056-CUM Rev. 1 dated 01.02.2023 SEE EASA APPROVED MODEL LIST for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Supported by Certification Programme C0055-CEP and C0056-CEP - per Certification Programme C0055-CEP and C0036-CUM, and C0056-CUM which are only one stretcher per aisle for Airbus Aircraft Numbers as stated in Certification Programme C0055-CEP and C0036-CEP. For additional aircra...</td>
<td>09/06/2017</td>
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<tr>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AEROTEC &amp; AIRBUS S.A.S.</td>
<td>Installation of a KANNAD 406 AF-H or KANNAD 406 AF-H Emergency Locator Transmitter.</td>
<td>Flight Manual Supplement, Doc. No. HAU-D-032-005, Rev. 0.; Instructions for Continued Airworthiness, Doc. No. HAU-D-032-006, Rev. 0.; later revisions of the above listed documents approved by EASA - Master Document List, Doc. No. HAU-D-032-001, Rev. 0.; Certification Plan, Doc. No. HAU-D-032-002, Rev. 0.; ...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>12/06/2017</td>
<td>Active</td>
<td>10042143</td>
<td></td>
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<tr>
<td>VECTOR AEROSPACE HELICOPTER SERVICES INC.</td>
<td>Installation of the L-3 Aviation Products Lynx multilink surveillance system model NGT-9000( ) part number 9029000-20000 or Garmin GTX-345ADS-B transponder part number 011-03302-40.</td>
<td>1. Master Drawing List (MOL) VAHS-1614-MDL, Revision NC, dated 25 November 2016, or later FAA approved revision.2. Instructions for Continued Airworthiness (ICA) for the L-3 Aviation Products Lynx Multilink Surveillance System Model NGT-9000( ) part number 9029000-20000 and Garmin GTX-345 ADS-B Transponder...</td>
<td>09/06/2017</td>
<td>Active</td>
<td>10042156</td>
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<tr>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>Installation of a KANNA D 406 AF-H Emergency Locator Transmitter.</td>
<td>Flight Manual Supplement, Doc. No. HAU-D-032-005, Rev. 0.; Instructions for Continued Airworthiness, Doc. No. HAU-D-032-006, Rev. 0.; later revisions of the above listed documents approved by EASA - Master Document List, Doc. No. HAU-D-032-001, Rev. 0.; Certification Plan, Doc. No. HAU-D-032-002, Rev. 0.; ...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>18/12/2017</td>
<td>Active</td>
<td>10042160</td>
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<td>AIRBUS S.A.S.</td>
<td>Installation of a Garmin GTN625 and approval for P-RNAV and LNAV/VNAV capabilities.</td>
<td>Installation in accordance with AEROTEC Concept modification ref AE1643 has defined in Master Data List ref AE1643MDL, Rev 0., dated 01/06/2017 and Service Bulletin ref AE16243SB, Iss 4 dated 19/05/2017. Operation in accordance with AEROTEC Concept Flight Manual Supplement ref AE16243FM, Rev 0., dated...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>12/06/2017</td>
<td>Active</td>
<td>10042171</td>
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<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAAC STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>SPAES GmbH &amp; Co. KG</td>
<td>REV. 1</td>
<td></td>
<td>PA-46-330P</td>
<td>Installation of Directional Antenna NY-164.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>- Certification Plan (Compliance Checklist)</td>
<td>CCL-046-066-53-01, Rev. 00</td>
<td>14/04/2017</td>
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<td>SA226-T, SA226-AT, SA226-TC, SFA41, SA226-TC, SA226-AT, SA226-T</td>
<td>MT-27-() on MT Metro II / Martin IV equipped with TPE331()-() engines</td>
<td>The Type and OSiD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: CS-38, Amendment 4AFR 36, Amendment 36-30. The requirements for environmental protection and the associated...</td>
<td>- According to MT-Propeller Master Documentation No. E-3056, Rev. 7. To be operated in accordance with MT-PropellerAFM-S Document No.E-3047, Rev. 3. To be installed i.a w. Installation Instruction Doc. No. E-3048, Rev. 0. To be maintained i.a w. MT-Propeller ICA Document No. E-3049, Rev 0 or later...</td>
<td>- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>14/12/2021</td>
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<tr>
<td>ARMSTRONG ENTEKWICKLING GmbH</td>
<td>REV. 1</td>
<td></td>
<td>737-800, 737-900(S/73-900ER, 737-800/737-800/737-900)</td>
<td>Certification of Compatibility with Cellular and Wireless DataTechnologies.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval as amended by the following additional or alternative airworthiness requirements:14 CFR Part 25 Regulations at the latest amendment 25-0 through 25-11925.109000, 25-1520. The requirements for enviro...</td>
<td>1. Prior to installation of this design change it must be determinedthat the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>30/06/2017</td>
<td>Active</td>
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<tr>
<td>EOXEL DESIGN LIMITED</td>
<td>REV. 4</td>
<td></td>
<td>B200CT, B200T 200, 200C, 200CT, 200T</td>
<td>Leading Edge Application Post (LEAP) Series – (LEAP – R, R 15%SB, Expanded) and speed(s) &lt;(v)-&lt;(v-1)-&lt;(v-2)-&lt;(v-3)-&lt;(v-4) Alternative Material</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval as amended by the following paragraph(s) at a later amendment: CS-23 Amendment 1.23.301,23.303, 23.305, 23.307, 23.573, 23.601, 23.603, 23.605, 23.613, 23.619, 23.625. The requirements for environmental protection and the associated...</td>
<td>For aircraft modified with the Post “1LEAP” - To be approved according to AFM supplement 2XL023LEF79(1) Issue 6 dated 18th April 2019 – To be operated according to AFM as amended by FAA approval revision.</td>
<td>As indicated in LEF79, Issue 8, dated 24th November 2020. Prior to installation of this design change/repair it must be determined that the interrelationship between this design change/repair and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>26/11/2020</td>
</tr>
<tr>
<td>SABENA TECHNICS BOD</td>
<td>REV. 1</td>
<td></td>
<td>A330-323</td>
<td>A330 Air Carabes – Ka-band system installation This modification consists in...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- Master Data List, ref. MDL-5-05-4309 issue A, dated 30 June 2017 - Instruction for Continued Airworthiness, ref. (ICA- F3046-00-010 issueB, dated 30 May 2017 - Aircraft Flight Manual Supplement, ref. AFMS- F3046-00-022 revision.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>14/03/2019</td>
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<tr>
<td>SABENA TECHNICS BOD</td>
<td></td>
<td></td>
<td>A330-323</td>
<td>A330 Air Carabes cabin and system modification. This modification consists in...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- Master Data List, ref. MDL-5-05-4307 issue B, dated 08 June 2017 - Aircraft Flight Manual Supplement, ref. AFMS- F3045-00-031 revision 1. dated 25 May 2017 - Instruction for Continued Airworthiness, ref. ICA- F3045-00-030Revision A, dated 28 May 2017 or later revisions of the above listed documents a...</td>
<td>The following modifications must be concurrently embodied with the modifications indicated below: - STC 10014507 Revision 2 – Floor Pathmarking installation - MOD- S-23-4087 AAP – Piacard modification (Minor change – Approval numbering DOA EASA 211.05(1) – 1056L-25-06 revision 00 - Gallely O12 modification (Minor change -Appr...</td>
<td>14/06/2017</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>SELL GmbH</td>
<td>REV 2</td>
<td></td>
<td>777-200LR, 777-300ER</td>
<td>Relocation of Galleys M1, M2, Petrel and BC Coat Stowage Rev.1. Aircraft Serial Numbers extension Rev.2. Type Definition Document update.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associatedtered noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Type Certification Document, SELL GmbH document No. TOD-10410-26-34(1)</td>
<td>OSID data must be approved before entry into service with an European Union operator. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise adverse effect upon.</td>
<td>15/08/2018</td>
</tr>
<tr>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>MOD. MD083 FALCON 900EX (F900EX) WITH</td>
<td>FANS 1A/OCL ENABLE switch installation in cockpit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associatedtered noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- DFS AFM Supplement Ref. F9000-037 dated 09 June 2017. MASTER DOCUMENT LIST Rev. DFS-06-17-F9-0826 Issue A dated 09 June 2017 or later revisions of the above listed documents approved by EASA.</td>
<td>Installation of Iridium Satcom, Immarsat Satcom and FANS 1A to be deactivated over Irdium are pre-requisite for this design change. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise adverse effect upon.</td>
<td>14/06/2017</td>
<td>Active</td>
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<tr>
<td>BOMBARDIER Inc.</td>
<td></td>
<td>TCCA SA17-28 ISSUE 1</td>
<td>BD-100-1A10</td>
<td>EASA validation of Transport Canada Supplemental Type Certificate SA17-28 at Issue 1. Replacement of DCMY Hydraulic Fluidline. Modification to be completed in accordance with B/E Aerospace Inc. Modification Data Summary number K922000 Revision NC or later approved revisions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associatedtered noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Modification to be completed in accordance with B/E Aerospace Inc. Modification Data Summary number K922000 Revision NC. For Instructions for Continued Airworthiness, refer to K922070 Revision NC, or later revisions of the above listed documents approved by EASA.</td>
<td>See those of original Transport Canada approved STC. SA17-28. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise adverse effect upon the airworthiness of...</td>
<td>16/06/2017</td>
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<tr>
<td>AMAC AEROSPACEx SWITZERLAND AG</td>
<td></td>
<td></td>
<td>777-200LR</td>
<td>Interior Outfitting</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements Special Condition (s)(D-02, Isolated Component,D-06, Glass Screens of Displays / Monitors,D-09, Shower Installation,D-15, Crew Rest Compartment and Flight Crew Rest Compartment...</td>
<td>- Master Data List MDL-DW01-2500 Rev. D dated 16.06.2017; Airplane Flight Manual Supplement AFMS-DW01-2500 Rev. C dated 02.06.2017; Instructions for Continued Airworthiness ICA-DW01-2500 Rev. B dated 14.06.2017. or later revisions of the above listed documents approved by EASA.</td>
<td>Per Airplane Flight Manual Supplement AFMS-DW01-2500; Per Instructions for Continued Airworthiness ICA-DW01-2500; Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise adverse effect upon the airworthiness of...</td>
<td>16/06/2017</td>
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<tr>
<td>AVIATION TECHNICAL SERVICES, Inc.</td>
<td>FAA STC</td>
<td>ST025418E</td>
<td>A319-131-132-133</td>
<td>Installation of a reconfiguration of the cabin interior</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval except where amended as listed below: Regulation 25.751 (c)(b), Amdt. 25-194; Regulation 25.866(a)(1)(4), Amdt. 25-112; Regulation 25.853(c)(i)(ii)(a)/(x)/(y)/(z), Amdt. 25-116; Regulation 25.854, Amdt. 5-149; Regulation...</td>
<td>1. Aviation Technical Services Master Data List (MDL) 721546- MDL-002, Revision J dated 28 April 2017, or later Federal Aviation Administration (FAA) approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise adverse effect upon the airworthiness of...</td>
<td>19/06/2017</td>
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<tr>
<td>AVIATION TECHNICAL SERVICES, Inc.</td>
<td>FAA STC</td>
<td>ST025268E</td>
<td>737-800, 737-900, 737-800, 737-900, 737-700</td>
<td>Installation of a Cargo Door Latch</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval except where amended as listed below: 14 CFR §21.50 Amdt. 21-9214 CFR §25.611(a) Amdt. 25-12314 CFR §25.783(a) (1)(x)(ii)(i)(iv)(ii)(xii) Amdt. 25-1144 CFR §25.899 Amdt. 25-12314 CFR §25...</td>
<td>Aviation Technical Services Master Drawing List (MDL)-718932- MDL-001, Revision K dated 11 January 2017, or later Federal Aviation Administration (FAA) approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise adverse effect upon the airworthiness of...</td>
<td>19/06/2017</td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10062277</td>
<td>WIPARE Inc.</td>
<td>FAA SA0323CH</td>
<td>182S, 182T</td>
<td>Installation of Lycoming IO-580-B1A engines and Herzltein-FYXR-WAN NG330I-3 Propllet at 3158 lbs Gross Weight on Textron Aviation Inc. model 182S and 182T.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this STC certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>- Wipare Helicopter Installation Instruction. Document No. 1005680, Revision F, dated January 1, 2016; - FAA Approved Airplane Flight Manual Supplement, Document No. POHSA0323CH-1.</td>
<td>1. This STC is currently valid for 182 planes only. 2. Engine limitations refer to Installation Instruction, Document No. 1005680</td>
<td>19/06/2017</td>
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<tr>
<td>10062279</td>
<td>BOGGI S.R.L.</td>
<td>AW169</td>
<td></td>
<td>Incubator Installation on AW169 Helicopter.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 29 amdt 3, 29-21, 29-25, 29-27, 29-29, 29-31, 29-251, 29-301, 29-305, 29-307, 29-309, 29-321, 29-337, 29-339, 29-341.</td>
<td>Rotorcraft Flight Manual Supplement 169PO0200-RFMS issue 1c,&lt;1&gt; or later revisions of the above listed documents approved by EASA Master Document List 169PG200-Mll issue 2, dated 14/06/2017. Declaration of compliance 169P0020-DoC issue 0, dated 14/06/2017.</td>
<td>This STC must be installed in conjunction with EASA STCs. 1058120000 P/N 169-040200-0501. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/06/2017</td>
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<td>10062280</td>
<td>PMV ENGINEERING</td>
<td>REV. 2</td>
<td>A320-211/212-231/232 A320-211/212-233 A320-211/212-214/215-218 A319-111/120-132 A319-111/120-113-114/-115</td>
<td>ELT survival installation on A320 family (A318 excluded) initial approval: the STC consists in installing a TECHTEST LIMITED EMERGENCY LOCATOR TRANSMITTER ELT (Type) Series 500-12 in an adequate overhead bin of the A320 family aircraft (Revision 1, a second configuration is added - Cockpit coat colors).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this STC certificate/approval. The following paragraph(s) at a later amendment: CS 29 amdt 18 for the following requirements: S61 (c)(1)(v)(b)(2)(d), 601, 611, 787(a)(1)(x)(2)(b), 789 (a), M63(a)(1)(x)(1)(x)(1)h(2)(c)(1), 150(a).</td>
<td>- MASTER DOCUMENT LIST, ref: AV-01510-MDL-1016 dated 05/03/2020 - INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, ref: AV-01510-ICA-R03 dated 14/11/2019 or later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/03/2020</td>
</tr>
<tr>
<td>10062281</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CO LTD (AIEC)</td>
<td>200</td>
<td></td>
<td>HySpec Camera replacement by Trimble A605 Camera System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this STC certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- To be installed according to Engineering Order No. CV5465-062, Rev. R dated 6 December 2016; To be operated according to AFM Supplement No. CV5465-071, Rev. R dated 15 December 2016; To be maintained according to ICA No. CV5465-085, Rev. 2, dated 11/February 2017 or later revisions of the above listed</td>
<td>Limited to the S/N: BB333. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/06/2017</td>
</tr>
<tr>
<td>10062286</td>
<td>AIRCRAFT SPECIALTIES SERVICES, Inc.</td>
<td>FAA STC S0090686C</td>
<td></td>
<td>Installation of SS39800M50 and SS39800M40 Starter Adapter Springs</td>
<td>The Certification Basis (CB) for the original product remains applicable to this STC certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Aircraft Specialties Services, Inc. STC Document List 1101A, Rev. 1 dated 3 May 2006, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/06/2017</td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC No.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10062288</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>REV. 1</td>
<td>FAA STD-SR0928RC-D</td>
<td>MBB-BK117 D-3 MBB-BK117 D-2 MBB-BK117 C-2</td>
<td>Installation of a Concorde RG-381E44 lead acid battery Revision 1 adds battery installation for the D3 model.</td>
<td>The Certification Basis for the original product remains applicable with this certification approval, except where amended by additional or later amendments if indicated on FAA STC SR0928RC-D. The requirements for environmental protection and the associated certified noise/ or emissions levels of the...</td>
<td>Drawing List 145AS29000, Revision E, dated 22 August 2012 Model MBB-BK 117 C-2. Rotorcraft Flight Manual Supplement 117FNS-221, Revision B, dated 8May 2013. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. Additional limitations may be detailed in the R...</td>
<td>04/02/2022</td>
</tr>
<tr>
<td>10062295</td>
<td>SAFRAN ELECTRONICS &amp; DEFENSEBUSINESS UNIT - HELICOPTERS DATAMANAGEMENT SERVICES &amp; SERVICES</td>
<td>REV. 1</td>
<td>412, 412EP</td>
<td>Installation of Flight Data Monitoring system HELICOM V2 ANA or HELICOM V2 ANA (PN4702F-ex) including (i) battery(PN46002) and Rack HELICOM V2ANA (PN4771-xx) or - Device HELICOM V2 ANA (PN4725P-xx) - PN4730P-xx)ANA (PN4...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s): CRI F-01 (Lithium Battery Installations). The requirements for environmental protection and the associated certified noise/ or emissions levels of the...</td>
<td>- RFNS. Doc. No. N111-BELL412 HV2 HV2+ ANA-E-A-03 Rev.3, dated20.12.2016(&gt;)(&gt;) in later revisions of the above listed documents approved by EASA. - Master Data List: Doc. No. N10-BELL412 HV2 HV2+ ANA-E-A-09, dated16.06.2017. - Installation Manual: Doc.No. N7-BELL412 HV2 HV2+ ANA-E-A-09, dated16.06.2017...</td>
<td>- This STC is not applicable to Model 412EP helicopters serial numbers 37102 thru 37999 with FADEC. - The system does not replace standard maintenance records and pilot reporting. Prior to installation of this design change it must be determined that the interrelationship between this design change and any...</td>
<td>27/02/2020</td>
</tr>
<tr>
<td>10062301</td>
<td>JET AVIATION AG</td>
<td>REV. 2</td>
<td>737-700ER</td>
<td>Live Live TV Antenna Structural Provisional Change to EASA STC 10062301 to remove the limitation of 500 FC ori months.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s): CRI C-102 Vibration &amp; buffeting compliance criteria for large external antenna installation. The requirements for environmental protection and...</td>
<td>JBSL-44-00514 KCA-09, Issue 0 Instruction for Cont. AirworthinessJBSL-M-00914 MCL-01, issue 10 Master Data Listter later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>13/12/2017</td>
</tr>
<tr>
<td>10062317</td>
<td>AIRTRONIC SAS</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>DR Series A/C Poly-Fiber Fabric</td>
<td>Poly-Fiber STC Aircraft Model List for DR Series Aircraft. Issue Date 01March 2017. Issue 01 dated 15-03-2017Installation and maintenance instructions “Supplement manual: maintenance – entoilage Poly-FIBER – avions DR220, DR300-400” Edition 01 dated 07-06-2017 or later revisions of the above listed d...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s): CRI C-102 Vibration &amp; buffeting compliance criteria for large external antenna installation. The requirements for environmental protection and...</td>
<td>None</td>
<td>23/06/2017</td>
<td>Active</td>
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<tr>
<td>10062336</td>
<td>BOMBARDIER Inc.</td>
<td>TCCA STC SAS-60 ISSUE 1</td>
<td>BD-700-1A11</td>
<td>Complete Custom Aircraft Interior.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements Special Condition (s):D-200 EASA Special Condition and Deviation. Side facing divans. Equivalent Safety Finding (s):D-500 Emergency exit signs reduction in size of lettering...</td>
<td>Aero Consulting Services Ltd. Configuration Definition ListE025200Revision FY dated 23rd March 2017, andAircraft Flight Manual Supplement G- FM25004501 Revision N/C or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU...</td>
<td>- See those of original Transport Canada STC SAS-60 and of the AFSupplement quoted above in the associated technical documentation para...</td>
<td>26/06/2017</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10062337</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>747-400</td>
<td>IFE WiFi system installation and activation</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: JAR 25 Change 14 for 25.853, 25.869, 25.1423 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product...</td>
<td>- Instruction for Centread Airworthiness ref: ICA: B747-16-002-001 Rev0 AFM ref: AFMS- B747-16-002-001 or later revisions of the above listed documents approved by EASA - Master Data List Change Approval Sheet ref: CAS-B747-16-002 Rev 0</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce roadwes effect upon the airworthiness of the product.</td>
<td>26/06/2017</td>
<td>Active</td>
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<tr>
<td>10062339</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>PA-28-236 (DAKOTA)</td>
<td>MTV-4 on Piper PA-28-236</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CS-36, Amendment 4FA 36, Amendment 36-207 This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or...</td>
<td>According to MT-Propeller Master Data List No. E-2878, Rev 1 To be operated in accordance with MT-Propeller AFM-S Document No. E-2885, Rev 1 or later revisions of the above listed documents approved by EASA. To be installed and/or Installation Instruction Doc. No. E- 2886, Rev 1 or later revision.</td>
<td>According limitation of AFM-S Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/06/2017</td>
<td>Active</td>
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<tr>
<td>10062340</td>
<td>BS DESIGNS, INC.</td>
<td>F3A STC SA09872AC</td>
<td>208/ 208B</td>
<td>Installation of Hartzell Propeller, Inc. Composite Propeller HC-435JNCL-1004(&gt;h) to De ice, or HC- 435JNCL-1004. Non-Delice.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and unchanged and remain applicable to this certificate/approval.</td>
<td>BS Designs, Master Data List BSD-MLD-058-01, Rev. B1, dated April 26, 2017 Airplane Flight Manual Supplement dated April 11, 2017 or later revisions of the above listed documents approved by EASA. EASA ED Decision 2004/04/CF or subsequent revisions of this decision and/or the Technical Instructions.</td>
<td>1. The installation of the Honeywell International Inc. TPE331-12JR-74HT turboprop engine is required, in accordance with EASA STC SA10841SC. 2. The Hartzell MC-435JNCL-1004(&gt;h) propeller diameter limits are 279.4 – 277.9 cm (110 to 105.9 inches). 3. Honeywell Service Bulletin No1...</td>
<td>23/08/2017</td>
</tr>
<tr>
<td>10062360</td>
<td>RUAG AG</td>
<td>REV. 1</td>
<td>DORNIER 228-202</td>
<td>Approval of Avionics Upgrade DOR228-202 Serial 8112.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Serial 8112 Only. Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/02/2021</td>
<td>Active</td>
</tr>
<tr>
<td>10062364</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>CL-600-2B16 (604)</td>
<td>VARIANT</td>
<td>Exterior LED Lights Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>QCM design GmbH Certification Report No. B605-QCM33-001 Initial Issue Dated 28th June 2017 Restricted to ASN 5301 to 5699 and ASN 5701 to 5699</td>
<td>27/06/2017</td>
<td>Active</td>
</tr>
<tr>
<td>10062368</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 2</td>
<td>CAA-UK RESPONSIBIL ITY</td>
<td>A319-111, A320-214</td>
<td>Activation of FeiTan System Media Server with Cabin Wi-Fi. Rev 1: Transfer from STC Twenty One to JAMCO Aero Design &amp; Engineering Rev. 2: Transfer from JAMCO Aero Design &amp; Engineering to STC Twenty One Limited.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>STC Twenty One Ltd Flight Manual Supplement No. S21-FMS-0513 Issue 1. STC Twenty One Ltd Master Data List No. S21-D1-1131 Issue 1 or later revisions of the above listed documents approved or accepted under the EASA system.</td>
<td>The design change only provides approval for aircraft tolerance to 4Ghz, 5Ghz and Cellular Rf radiation and the operation of the multiuser with camera Wi-Fi. Portable Electrons Devices (PEDs) must still be operated as per the aircraft PED Policy. It is the responsibility of the Aircraft Operator to...</td>
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<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>P206T</td>
<td>STC-AT-108 Noise Certification</td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection and a change to the certified noise or emissions levels.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>29/06/2017</td>
<td>Active</td>
</tr>
<tr>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>CA-UK RESPONSIBIL</td>
<td>MBB-BK117 D-2</td>
<td>Full NVIS certification.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>30/06/2017</td>
<td>Active</td>
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<tr>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 1</td>
<td>A330-203</td>
<td>A330-203 Cabin Modification Rev.1.- Transfer of STC from SR Technics Switzerland to gROWING Engineering Partners GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/04/2021</td>
<td>Active</td>
</tr>
<tr>
<td>KUERZI AVIONICS AG</td>
<td>AS 350 B2, AS 350 B3</td>
<td>Installation of EFIS IDU450 and general avionics upgrade of AS350 helicopters.</td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/06/2017</td>
<td>Active</td>
</tr>
<tr>
<td>GFD GmbH</td>
<td>REV. 1</td>
<td>35A</td>
<td>Iridium Satcom Antenna Installation Provision Rev. 1- Transfer to GFD GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Applicable to SN: 35-654. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
<td>Active</td>
</tr>
<tr>
<td>CHIPPED AEROSPACE, Inc.</td>
<td>FAA STC STA4367AT</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Removal of the Existing Honeywell DL-400 or DL-400 Data Loader and Installation of the Honeywell DL-1000 Data Loader.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. The STC is valid for those aircraft types...</td>
<td>03/07/2017</td>
<td>Active</td>
</tr>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>FOKKER SERVICES B.V.</td>
<td>DHC-8-201, DHC-8-202</td>
<td>ADS-B out on the Dash 8-200 series</td>
<td>- DHC8-FX1-AFMS-004, Airplane Flight Manual Supplement, Revision 1 dated 20 Jan 2017, DHC8-FX1-MMEL-S-001, Master Minimum Equipment List Supplement 1 dates 11 Apr 2017</td>
<td>Prior to installation of this design change must be determined that the relationship between this design change and any other previously installed design change or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/07/2017</td>
<td>Active</td>
<td>10202491</td>
<td></td>
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<tr>
<td>TASS-EU Ltd.</td>
<td>787-300</td>
<td>Installation of EROS Full Face Quick Donning Mask Regulator M40-40-007A</td>
<td>TASS-EU Flight Manual Supplement No. TASS-004-007A</td>
<td>None</td>
<td>05/07/2017</td>
<td>Active</td>
<td>10202538</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>EC135 P2+</td>
<td>Laser illumination – MX 15HD camera.</td>
<td>Flight Manual Supplement STC-111-0-AFM for Rotorcraft EC135P2+ “Laser Illuminator Activation”, Rev. 3, dated 14 July 2017</td>
<td>This STC is limited to EC135 s/n 660 Requires as a precondition the installation of: EASA STC 10040209, Rev. 2 - EASA Minor Change 1005998. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/07/2017</td>
<td>Active</td>
<td>10202521</td>
<td></td>
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<tr>
<td>AVIANOR INC</td>
<td>BD-500-1A11</td>
<td>Design and Installation of Removable Seat Table.</td>
<td>Avianor Master Document List No. MDL-16150701, Revision F dated 13 March 2019</td>
<td>Restricted to ASNs 55003, 55004, 55005, 55006, 55007, 55008, 55009, 55016, 55034, 55035, 55036, 55041, 55042, 55043, 55050, 55051, 55052, 55053, 55054, 55055, 55056, 55057, 55058, 55059. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and...</td>
<td>09/07/2019</td>
<td>Active</td>
<td>10202524</td>
<td></td>
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<tr>
<td>DART AEROSPACE Ltd.</td>
<td>R66</td>
<td>Dart Aerospace Ltd. Hel-Utility-Basket Installation for Robinson R66</td>
<td>Flight Manual Supplement Hel-Utility-Basket Robinson R66 FMS-019 Issue 1, Revision A.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/07/2017</td>
<td>Active</td>
<td>10202495</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA no.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10062531</td>
<td>FLYING COLOURS CORP.</td>
<td></td>
<td></td>
<td>CL 600-281E (604 VARIANT)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>B/E Aerospace Airplane Flight Manual Supplement No. K179000Rev N/C(&gt;)+ &gt;no later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU/Canada/Bilateral Agreement/B/E Aerospace Modification Data Summary No. K179000 Rev N/C.</td>
<td>Restricted to ASN 5520.</td>
<td>08/12/2017</td>
</tr>
<tr>
<td>10062542</td>
<td>COMMUTER AIR TECHNOLOGY Inc.</td>
<td></td>
<td>FAA STC SA69555W</td>
<td>B200T, B200GT, B200GT B200, B200C, B200C A101-1, A200, A200C, A200CT 300, 300LW, B300, B300C 200, 200C, 200CT, 200T</td>
<td>Installation of full turning contoured engine exhaust stacks</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>1. Drawing No. DL19505, Revision 3, or later FAA approved revision 2.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10062544</td>
<td>BABCOCK MISSION CRITICAL SERVICES DESIGN AND COMPLETIONS LIMITED</td>
<td>REV. 2</td>
<td>CAA-UK RESPONSIBILITY</td>
<td>EC135 T2, EC135 T2+; EC135 T3; EC135 P1; EC135 P2, EC135 P2+</td>
<td>TPED Tolerance Certification on EC135Revision 1: Extension of STC applicability to EC135 P2, EC135 P2+, EC135P1, EC135 P2; EC135 P2+</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- Roskco manufactured Supplement FMS/140-002 Issue 4 - Instruction for Continued Airworthiness ICA/140-007 Issue 5 - Design Report DR-140-001 Issue 7 - Post Installation Actions ICA/140-008 Issue 3 - Modification Leaflet &amp; Certificate of Design Issue 4or later revisions of the above listed document(s) apply.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10062546</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td></td>
<td>55/ 55B: 56C</td>
<td>Replacement of DFDR F800 by DFDR F1500.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for Special Condition (a) CRI-F-01 Non-Chargeable Li-On Bat. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unaltered and remain applicable to this certificate/ approval.</td>
<td>According to Aero-Dienst Master Document List No. 20299-LJ.DC-101, Rev. 1.</td>
<td>According to Aero-Dienst SMA Document No. 20284-LJ.DC-106 Rev. 00 or later revisions of the above listed documents approved by EASA. To be installed i.e. Installation Instruction Doc. No. 20299-LJ.DC-104 T.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC/Issue</td>
<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10062566</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>FAA STC</td>
<td>ST04143CH</td>
<td>737-900, 737-900ER</td>
<td>EASA validation of FAA STC ST04143CH. Installation of a Telephone/Entertainment (T/Ent) System in accordance with Autonics Armstrong Aerospace Master Data List, Doc. No. DC104-9734-00, Revision (J) dated 15th May 2017 or later approved revisions.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>- Flight Manual Supplement P2006T-HK007 issue for later revisions of the above listed documents approved by EASA. Modification description D 200676-DE501 Ed. 4</td>
<td>10/07/2017</td>
</tr>
<tr>
<td>10062569</td>
<td>ARMSTRONG AEROSPACE INC.dba ASTRONICS ARMSTRONG AEROSPACE</td>
<td>FAA STC</td>
<td>ST04143CH</td>
<td>737-700, 737-800</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment CS25 impacted requirements at initial issue:&lt;(&gt;,&lt;) &gt; unless otherwise specified: 25.29, 25.301(a), (b)(amdt 1); 25.303, 25.305(a)...</td>
<td>Certification Basis for this STC to be performed in accordance with Autonics Armstrong Aerospace Master Data List, Doc. No. DC104-9734-00, Revision (J) dated 15 May 2017, and The Airplane Flight Manual Supplement, Autonics Armstrong Aerospace Doc. No. DC303-9734-00, Rev (-) approved 07th October 2016 or later.</td>
<td>1) See those of original FAA STC ST04143CH and those of the later revisions of the above listed documents approved by EASA. Modification description D 200676-DE501 Ed. 4</td>
<td>10/07/2017</td>
</tr>
<tr>
<td>10062574</td>
<td>KAMAN CORPORATION ROCRAT WHEEL AND BRAKE, LLC</td>
<td>FAA STC</td>
<td>SA781GL</td>
<td>404</td>
<td>Replace existing Main Wheel and Brake by installing ParkerHannifinConversion Kit 199-78.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certification approval. The requirement for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>Installation Instructions and Installation Drawing 55-54. Rev. C dated 31May 1984 or later FAA-approved revision.</td>
<td>10/07/2017</td>
</tr>
<tr>
<td>10062577</td>
<td>ALPINE AEROTECH LP</td>
<td>REV. 1</td>
<td>TCCA SH13-13</td>
<td>ISSUE 1</td>
<td>Aft Cabin Wall Protection Kit (TCCA STC SH13-13, Issue 1) Installation of polymer panels on top of the cabin pylon and aft cabin composite panel walls, to provide protection from damages and wear. Polymer panels are bolted on inserts already installed on the nose composite panels.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certification approval. Requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>AAL-212-030-701, Revision A, dated 2015-09-29 AFT CABIN BULKHEAD PROTECTOR, KIT INSTRUCTIONS FOR CONTINUING AIRWORTHINESS P2006T Serial Number 085 &amp; 129 Prior to installation of this design change it must be determined whether the interrelationship between this design change and other previously installed design change and/or repair will introduce roadvane effect upon the airworthiness of the...</td>
<td>25/07/2017</td>
</tr>
<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10062579</td>
<td>GOERRES, HARALD</td>
<td>REV. 1</td>
<td>STC Holder</td>
<td>FOURNIER RF S B &quot;SPERBER&quot; FOURNIER RF S</td>
<td>Various Improvements reg. instruments.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Modification Bulletin (Major Change HD-MC-RFS-001c later revisions of the above listed documents approved by EASA-Erklärungen zum Flugleistungsbeispiel der Änderung des Instrumenten/Panel/ Supplemental pages to Flight Manual, relating to modification of the instrument panels) dated May 2017.</td>
<td>19/07/2017</td>
</tr>
<tr>
<td>10062583</td>
<td>21JAERODESIGN LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A320-25ZN-26ZN</td>
<td>Introductions of Zero-passerenger Seating Configuration/revision: Extension of applicability</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraphs at a later amendment: CS.25 Amendment 18 - CS.25.26, CS.25.27, CS.25.601(C&gt;)(&gt;CS.25.785). CS.25.803, CS.25.813, CS.25.815, CS.25.817, CS.25.853. CS.</td>
<td>21Jaerodesign Design Change Document No. DCG-007M Issue 021Jaerodesign Design Change Document No. DCG-007M Issue 01c later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>23/04/2019</td>
</tr>
<tr>
<td>10062588</td>
<td>AIRBUS S.A.S.</td>
<td></td>
<td>MOD C.2060</td>
<td>Installation of A350 2KU Antenna: installation of outside-aircraft equipment for 2Ku system including large antenna, base plate, adapter and lugs.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Vibration / buffeting compliance criteria for large/medium antenna installation. The requirements for environmental protection...</td>
<td>Certification Summary Report ref. A3C-JA-19015 Issue 6. Modification Approval Sheet: MAS C.2060-11 C. Instructions for Continued Airworthiness (CA). MD-P-305 C.2060Rev 101c later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined if this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/07/2017</td>
</tr>
<tr>
<td>10062593</td>
<td>HELIFAB Inc./AIRCRAFT AVIATION CO. LLC</td>
<td>FAA STC</td>
<td>S802507LA</td>
<td>206L-4, 407 206B, 206L-1, 206L-3</td>
<td>Installation of radar altimeter antenna mounting provisions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. HELIFAB Inc. Master Document List HF. MD-L-76, Revision IR. dated 28 January 2016, or later FAA approved revision 2. Continued Airworthiness. ICA. HF. CA-76, Revision IR. dated 25 January 2016, or other FAA accepted ICA.</td>
<td>Prior to installation of this design change it must be determined if this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10062601</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>FAA STC</td>
<td>ST02506LA</td>
<td>SEE EASA-APPROVED MODEL LIST</td>
<td>EASA Validation of FAA approved AML STC ST02506LA for the NKT-705S system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>80092603-001 Rev EASA Approved Airplane Flight Manual Supplement. 80092603-001 Rev. A Approved Amended MDL. 80092603-005 Rev. A. ICA. EASA AML List (attached to EASA STC) later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions).</td>
<td>Prior to installation of this design change it must be determined if this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10062607</td>
<td>SABENA TECHNICS BGC</td>
<td>FAA STC</td>
<td>SH02615H</td>
<td>F 406</td>
<td>Replacement of GPS for RNAV, LNAV and LPV capability on CESSNA F406.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS23.303, 23.305, 23.307(a), 23.311, 23.337, 23.341, 23.610(b) (2), 23.610a...</td>
<td>AFM Supplement Ref 0835-01-A-2301-R01 (12- Jul 2017)(&gt;)&gt;(&gt;)or later revisions of the above listed documents approved by EASA. ICA Ref. 0835-01-A-2211-R00 (11- Jul 2017)(&gt;)&gt;(&gt;)MDL Ref. 0835-01-A-2103-R10 (12- Jul 2017).</td>
<td>Prior to installation of this design change it must be determined if this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>STC Holder</td>
<td>STC No.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10062614</td>
<td>SFA SOLUTIONS FOR AVIATION, S.L.</td>
<td>60 ISAT SYSTEM AND ANTENNA INSTALLATION of SkyDataAirCom systems ISAT-100 and ISAT-200A (only onethat can be installed in the aircraft at the same time) in cockpit avionics bay; and related GPS/INU/antenna installation on top of fuselage with corresponding modification and reinforced...</td>
<td>The Certification Basis for the original product as amended threfollowing additional or alternative airworthiness requirements: Special Condition(s): CRI-F-01: Lithium battery installations. The requirements for environmental protection and the associated certified noise and/or emissions 1s of I...</td>
<td>None</td>
<td>13/07/2017</td>
<td>Active</td>
<td>10062614</td>
<td></td>
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<tr>
<td>10062621</td>
<td>PMV ENGINEERING</td>
<td>A310-304I-308I-322I-324I-325 A310-20I-203 C1204I-2204I-222 A300 F4-62 R A300 C4-620 F4-203I- F4-605 R A300 C4-203I C4-605 R VARIANT A300 B4-620I B4-603I- B4-605 R A300 B4-203I B4-220I- B4-605 I A300 B4-102I B4-103I- B4-120 A300 B2-203I B2-320I B2K-3C A300 B1I B2-1AI B2-1C2- B2-202</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchange...</td>
<td>None</td>
<td>14/07/2017</td>
<td>Active</td>
<td>10062621</td>
<td></td>
</tr>
<tr>
<td>10062622</td>
<td>IMAGIK INTERNATIONAL AIR REPAIR STATION, CORP.</td>
<td>REV. 1 FAA STC ST04365AT 737-700, 737-800 Installation of the IMAGIK QUATRO In- Seat Power Supply System (PSS); - Rev. 1: added the configuration 2 (Recaro seats P/N 5359/581/31-31) and 3 and configuration 3 (USB-C connection))</td>
<td>The Certification Basis for the original product as amended threfollowing additional or alternative airworthiness requirements: The following paragraphs at CS25 Amendment 4 CS25.23(a), CS25.301(a)(b), CS25.303, CS25.305(a)(b), CS25.581(c), CS25.601, CS25.603(a)(3)(k), CS.25.611, CS.25.613(a)(3)(k). C...</td>
<td>1. Master Data List (MDL) MD1002-01, Rev. N, dated 2 August 2022(1)(&lt;1). Instructions for Continued Airworthiness (CA) document CA1125-02 Rev.G, dated 30 June 2023, for configuration 1 document CA1220-01 Rev.C,dated 31 July 2023(&lt;1) -&gt;&gt; or later revisions of the above listed document(s) approved...</td>
<td>16/11/2023</td>
<td>Active</td>
<td>10062623</td>
<td></td>
</tr>
<tr>
<td>10062630</td>
<td>TURK HAVA YOLLARI TEKNIN A.S (TURKISH TECHNINC Inc.)</td>
<td>A321-231 Cabin RetrofitThis change covers a Cabin Reconfiguration with Introduction of NewSeats, Monuments and IFE System Installation according to the specification in the certification program C0018-CEP-06</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchange...</td>
<td>MDN 338 Prior to installation of this design change it must be determined that the...</td>
<td>17/07/2017</td>
<td>Active</td>
<td>10062630</td>
<td></td>
</tr>
<tr>
<td>10062644</td>
<td>DASSAULT FALCON SERVICE</td>
<td>FALCON 2000EX FALCON 2000 Modification DFS D220 Issue A - Baby sofa installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Master Document List DFS- DE-17-02 D220 Issues A dated 17 July 2017(&lt;1) -&gt;&gt; or later revisions of the above listed document(s) approved by EASA. Applicability is limited to...</td>
<td>18/07/2017</td>
<td>Active</td>
<td>10062644</td>
<td></td>
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<td>STC Holder</td>
<td>STC no.</td>
<td>STC Type</td>
<td>Foreign/NAACertification</td>
<td>Description</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>WORLDWIDE AVIATION L.L.C.</td>
<td>FAA STD 10062655</td>
<td>SAB1958W</td>
<td>Conversion from passenger to cargo only configuration for M7 AerospaASSS Metro series.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>According to FAA 87915WS and listed in AFM 6 Prior to installation of this design change it must be determined thatthe interaction between this design change and any otherwise previously installed design change and/ or repair will introduce noise/ adverse effect upon the airworthiness of the product.</td>
<td>19/07/2017</td>
<td>Active</td>
<td>100400205</td>
</tr>
<tr>
<td>SABENA TECHNICS BOC</td>
<td>10062657</td>
<td>ATR 72-212 A</td>
<td>The Certification Basis for the original product as amendedby the following additional or alternative airworthiness requirements: The following paragraph(s) is/are at a later amendment: CS-25 Amtd 4-Refer to EASA with Certification Plan, ref: ASB-01-A-CP-R03. Paragraph 6 for detailed additional requirement.</td>
<td>- MMEL, MMEL Supplement ref: 0828-01-A-CP-R02.- ASB-01-A-CP-R03 at later revisions of the above listed documents approved by EASA. Master Data List, ref: 0828-01-MMRL-R00.- Certification Plan, ref: 0828-01-A-CP-R03.</td>
<td>Prior to installation of this design change it must be determined thatthe interaction between this design change and any otherwise previously installed design change and/ or repair will introduce noise/adverse effect upon the airworthiness ofthe product.</td>
<td>19/07/2017</td>
<td>Active</td>
<td>100400205</td>
</tr>
<tr>
<td>TURK HAVA YOLLARI TEKNIK AS (TURKISH TECHNIC Inc.)</td>
<td>10062658</td>
<td>A340-542</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Certification Programme 8127-CEP-02.</td>
<td>Prior to installation of this design change it must be determined thatthe interaction between this design change and any otherwise previously installed design change and/ or repair will introduce noise/adverse effect upon the airworthiness of the product.</td>
<td>19/07/2017</td>
<td>Active</td>
<td>100400205</td>
</tr>
<tr>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>10062665</td>
<td>TCCA 00 or later revisions of the above listed documents approved by EASA. To be operated in accordance with AirAlliance Airworthiness, Revision 2.</td>
<td>ATR 42-200, ATR 42-300</td>
<td>Prior to installation of this design change it must be determined thatthe interaction between this design change and any otherwise previously installed design change and/ or repair will introduce noise/adverse effect upon the airworthiness of the product.</td>
<td>20/07/2017</td>
<td>Active</td>
<td>100400205</td>
<td></td>
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<tr>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td>10062689</td>
<td>DHC-8-314, DHC-8-315</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- MOD140050_AFMS_MED, Aircraft Dispatch Manual Supplement, Revision 1.0,05/06/2017.- MOD540000_ICA-MED, Instructions for Continued Airworthiness, Revision 2 dated 10/07/2017.- MOD540000_ICA-MED, D.O. Service Bulletin, Revision 1. dated 05/07/2017 or later revisions of the above listed documents approved.</td>
<td>Prior to installation of this design change it must be determined thatthe interaction between this design change and any otherwise previously installed design change and/ or repair will introduce noise/adverse effect upon the airworthiness of the product.</td>
<td>01/07/2017</td>
<td>Active</td>
<td>100400205</td>
</tr>
<tr>
<td>AIR ALLIANCE GmbH</td>
<td>10062690</td>
<td>F1000 SSFDR</td>
<td>The Certification Basis for the original product as amendedby the following additional or alternative airworthiness requirements: Special Additional Condition(s): (s)/CRI-F-01 Non-Chargeable LOBBar.</td>
<td>According to AirAlliance Master Document No. 2016_023_L35, Form 2.Rev 0 or later revisions of the above listed documents approved by EASA. To be installed a.c.r. Installation Instruction Doc. No. 2016_023_L35.</td>
<td>According to AFM-SA8915WS and listed in AFM 6 Prior to installation of this design change it must be determined thatthe interaction between this design change and any otherwise previously installed design change and/ or repair will introduce noise/adverse effect upon the airworthiness ofthe product.</td>
<td>21/07/2017</td>
<td>Active</td>
<td>100400205</td>
</tr>
<tr>
<td>STC Holder</td>
<td>System Installation of an Executive Interior and Installation of MKVIII EGPWS Limitations/Conditions 757-200, 757-300</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>TEXTRON, Inc. HELICOPTER</td>
<td></td>
<td>INSTALLATION OF AN EXECUTIVE INTERIOR AND INSTALLATION OF MKVIII EGPWS</td>
<td>THE CERTIFICATION BASIS (CB) FOR THE ORIGINAL PRODUCT REMAINS APPLICABLE TO THIS CERTIFICATE.APPROVAL. THE REQUIREMENTS FOR ENVIRONMENTAL PROTECTION AND THE ASSOCIATED CERTIFIED NOISE AND/OR EMISSIONS LEVELS OF THE ORIGINAL PRODUCT ARE UNCHANGED AND REMAIN APPLICABLE TO THIS CERTIFICATE/APPROVAL.</td>
<td>ACCORDING TO AIR ALIANCE MASTER DOCUMENT LIST NO. 2016_020_L335 FORM 2/REV 1 AND IN ACCORDANCE WITH AIR ALIANCE AIR CIR-050, Rev 05/06/2003 OR LATER REVISIONS OF THE ABOVE LISTED DOCUMENTS APPROVED BY EASA TO BE INSTALLED.</td>
<td>ACCORDING TO LIMITATION SECTION OF AFR-SPEC FOR INSTALLATION OF THIS DESIGN CHANGE IT MUST BE DETERMINED THAT THE INTERRELATIONSHIP BETWEEN THIS DESIGN CHANGE AND ANY OTHER PREVIOUSLY INSTALLED DESIGN CHANGE AND/OR REPAIR WILL INTRODUCE NO ADVERSE EFFECT UPON THE AIRWORTHINESS OF THE PRODUCT.</td>
<td>21/07/2017</td>
<td>Active</td>
<td>10062691</td>
</tr>
<tr>
<td>STG AEROSPACE GERMANY GmbH</td>
<td></td>
<td>INSTALLATION OF AN EXECUTIVE INTERIOR AND REDUCTION IN MAXIMUM ZERO FAIRWEIGHT FROM 28,814 TO 28,279 POUNDS</td>
<td>THE CERTIFICATION BASIS (CB) FOR THE ORIGINAL PRODUCT REMAINS APPLICABLE TO THIS CERTIFICATE/ APPROVAL. THE REQUIREMENTS FOR ENVIRONMENTAL PROTECTION AND THE ASSOCIATED CERTIFIED NOISE AND/OR EMISSIONS LEVELS OF THE ORIGINAL PRODUCT ARE UNCHANGED AND REMAIN APPLICABLE TO THIS CERTIFICATE/ APPROVAL.</td>
<td>1. DRB Aviation Consultants Master Document List Number DL163450 Revision D, dated 4 February 2009, or later FAA approved revision.</td>
<td>25/07/2017</td>
<td>Active</td>
<td>10062594</td>
<td></td>
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<tr>
<td>STG AEROSPACE GERMANY GmbH</td>
<td></td>
<td>MANUFACTURE OF WINDSHIELD UPGRADE/REVISION 1. TRANSFER OF STC HOLDER/pdf from STG AEROSPACE LIMITED TO STG AEROSPACEX GERMANY GMBH</td>
<td>THE CERTIFICATION BASIS (CB) FOR THE ORIGINAL PRODUCT AS AMENDED BY THE FOLLOWING ADDITIONAL OR ALTERNATIVE AIRWORTHINESS REQUIREMENTS: THE FOLLOWING PARAGRAPH(S) AT A LATER AMENDMENT/CS25 AT AMENDMENT 198/25.1, CS 25.25, CS 25.27, CS 25.301 (a)(ii), CS 25.303, CS 25.305 (b)(ii)(c), CS 25.307 (a), (d), (c),</td>
<td>1. MASTER DATA LIST, Doc No: SG-1015-MDL-01, Rev No: IR-6015-ICA-01, Rev No: IR-6015-ICA-01, REV NO: IR-6015-ICA-01, - SERVICE BULLETIN, Doc No: SG-1015-ICA-01, Rev No: IR-R 6015-ICA-01, REV No: IR-R 6015-ICA-01, leaving revisions of the above listed document(s) approved/accredited under the EASA system.</td>
<td>PRIOR TO INSTALLATION OF THIS DESIGN CHANGE IT MUST BE DETERMINED THAT THE INTERRELATIONSHIP BETWEEN THIS DESIGN CHANGE AND ANY OTHER PREVIOUSLY INSTALLED DESIGN CHANGE AND/OR REPAIR WILL INTRODUCE NO ADVERSE EFFECT UPON THE AIRWORTHINESS OF THE PRODUCT.2. APPLICABLE TO AIRCRAFT EQUIPPED WITH...</td>
<td>12/02/2021</td>
<td>Active</td>
<td>10062702</td>
</tr>
<tr>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td></td>
<td>INSTALLATION OF REPLACEMENT SKID TUBES THIS CERTIFICATE IS BASED ON &quot;GRANDFATHERED&quot; APPROVALS LBA RC1084, UKCAA 26707 AND 26708.</td>
<td>THE CERTIFICATION BASIS (CB) FOR THE ORIGINAL PRODUCT REMAINS APPLICABLE TO THIS CERTIFICATE/ APPROVAL. THE REQUIREMENTS FOR ENVIRONMENTAL PROTECTION AND THE ASSOCIATED CERTIFIED NOISE AND/OR EMISSIONS LEVELS OF THE ORIGINAL PRODUCT ARE UNCHANGED AND REMAIN APPLICABLE TO THIS CERTIFICATE/ APPROVAL.</td>
<td>1. Replacement of Windshield per Aeronautical Accessories, Inc. Drawing List Report No. AA-97002, Revision B, dated 12 June 1992 or later FAA approved revision.</td>
<td>PRIOR TO INSTALLATION OF THIS DESIGN CHANGE IT MUST BE DETERMINED THAT THE INTERRELATIONSHIP BETWEEN THIS DESIGN CHANGE AND ANY OTHER PREVIOUSLY INSTALLED DESIGN CHANGE AND/OR REPAIR WILL INTRODUCE NO ADVERSE EFFECT UPON THE AIRWORTHINESS OF THE PRODUCT.</td>
<td>17/06/2017</td>
<td>Active</td>
<td>10062707</td>
</tr>
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</table>

**Notes:**
- **EASA STC No.**: Unique identifier for each Supplemental Type Certificate (STC).
- **STC Holder**: The company or organization issuing the STC.
- **System**: The specific system or component being certified.
- **Rev. level**: Revision level of the STC or system.
- **Foreign/NAA STC ref.**: Reference to any foreign or non-European Aviation Authority Supplemental Type Certificate.
- **Models**: The models for which the STC is applicable.
- **Description**: A brief description of the STC.
- **Certification basis**: Details on the basis for the certification, including references to applicable regulations.
- **Associated Technical Documentation**: Any additional documentation required for the STC.
- **Limitations/Conditions**: Notes on any limitations or conditions applicable to the STC.
- **Date of Issue**: The date the STC was issued.
- **Status**: Whether the STC is active, expired, or revoked.
- **System Record**: Record number associated with the STC.
<table>
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<tr>
<th>STC Holder</th>
<th>Rev. Level</th>
<th>Foreign/NAATC Holder</th>
<th>Model</th>
<th>Description</th>
<th>Certification Basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>10062709</td>
<td>FAA STC SR00040AT</td>
<td>212, 412, 412EP</td>
<td>Installation of an extreem rainshield. This Certificate is based on the &quot;grandfathered&quot; approval ENAC99/255/MAE, UK CAA 26707 and 26708.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Aeromadaeous Accessories, Inc. Drawing list, Report No. AA-93085, Revision A dated 4 November 1993, and installation per Report No. 1993, Revision A dated 21 February 1994, or Report No. AA-97087, Revision A dated 21 May 1997, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>17/02/2017</td>
<td>Active</td>
<td>10062709</td>
</tr>
<tr>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>10062710</td>
<td>FAA STC SR00466AT</td>
<td>205A-1, 212, 412, 412EP</td>
<td>Manufacture of Filestop. This Certificate is based on the &quot;grandfathered&quot; approval LBA RC1134.ENAC 98/5476/MAE.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Aeromadaeous Accessories, Inc. Drawing List Report No. AA-93085, Revision A dated 4 November 1993, and installation per Report No. AA-93086, Revision A dated 21 February 1994, or Report No. AA-97087, Revision A dated 21 May 1997, or later FAA approved revision.</td>
<td>1. Inspection and Maintenance requirements are listed in Aeromadaeous Accessories, Inc. Report AA-98049, no revision, dated 32 December 1998 or later FAA approved revisions. 2. Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>17/06/2017</td>
<td>Active</td>
<td>10062711</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>10062713</td>
<td>P.68C</td>
<td></td>
<td>Title: STC-AT-113, Survey Cabin Configuration/Description: Installation of a flight guidance display in the cockpit, including computer devices, a laser scanner (Class 3B Laser) above the windshield, a photogrammetric hatch, an operator desk with data equipment (display, etc.), a Connector Box, a laser...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraphs of CS 23: Amendment 3: CS 23.29, CS 23.301, CS 23.305, CS 23.307, CS 23.337, CS 23.561, CS 23.601, CS 23.603, CS 23.605, CS 23.607, CS 23...</td>
<td>-Supplement to the AMM, Doc. No. STC-113-AFM, Rev. 1 or later revisions of the above listed documents approved by EASA, -Applicable Documents, (List of Compliance documents and manuals), Doc.No. STC-AT-113, Rev. 6 Supplement to the AMM, Doc. No. STC-113-AMM, Rev. 1. Installation Instructions, Doc...</td>
<td>Limited to S/N 496 Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>24/07/2017</td>
<td>Active</td>
<td>10062713</td>
</tr>
<tr>
<td>LB AIRCRAFT ENGINEERING LLC</td>
<td>10062719</td>
<td>FAA STC STG038AT</td>
<td>737-600-700-800-900</td>
<td>Installation of a Rockwell Collins CMU-900 ACARS/CPDLC Communication Management Unit P/N 822-1292-101 to provide Data Link communications evacuating data link capable VHF Mode 2 (VDLM2), HF, and/or SATCOM systems and the CPDLC application which support EASA Link 2000 communications Rev. 1 MMEL Sv...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraphs of CS 23: Amendment 3: CS 23.29, CS 23.301, CS 23.305, CS 23.307, CS 23.337, CS 23.561, CS 23.601, CS 23.603, CS 23.605, CS 23.607, CS 23...</td>
<td>-LB Aircraft Engineering Master Data List LB-CMU, 737-700-800-900, Rev. E dated 18 April 2017, or later FAA approved revisions; -Flight Manual Supplement, LB Aircraft Engineering document LB-CMU, 737-700-800-900, Rev. D dated 19 May 2017, or later FAA approved revisions; -LB Aircraft Engineering Instruction for Continued Operation...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>19/10/2020</td>
<td>Active</td>
<td>10062719</td>
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<tr>
<td>STG AEROSPACE GERMANY GmbH</td>
<td>10062724</td>
<td>REV. 1</td>
<td>CL-600-2D24 (RU SERIES 900) CL-600-2D15 (RU SERIES 705) CL-600-2C10 (RU SERIES 700,701,702) CL-600-2B10 (RU SERIES 100)</td>
<td>Installation of a Supertop Photoluminescent Aisle Marking System from STG Aerospace Limited to STG Aerospace GmbH</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraphs at a later amendment: CS25 AMENDMENT 18 Requirements: 25.1(a)-25.25 (20); 25.27, 25.301, 25.303; 25.305,25.307, 25.561(a)(1); 25.581; 25.60...</td>
<td>STG AEROSPACE LTD AEROPLANE FLIGHT MANUAL SUPPLEMENT No. SG-1004- FMS-890 Rev. 1 or later FAA approved revisions STG AEROSPACE LTD INSTRUCTIONS FOR CONTINUED AIRWORTHINESS No.SG-1004-ICA-01 Revision R STG AEROSPACE LTD CABIN CREW DATA SUPPLEMENT No. SG-1004-CCT-01 Revisioor STG AEROSPACE LTD MASTER MINIMUM EQUIPMENT LIST SUPPL...</td>
<td>SSIL Green Emitting Photoluminescent insert shall not be mixed with SSIL Blue Emitting Photoluminescent insert on the same aircraft. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>12/02/2021</td>
<td>Active</td>
<td>10062724</td>
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<tr>
<td>STC Holder</td>
<td>STC STC no.</td>
<td>Rev. level</td>
<td>Foreign/NA</td>
<td>STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>AEROTECH &amp; CONCEPT</td>
<td>REV. 1</td>
<td>A321-201N-2712N to A321-213I</td>
<td>ATR 72-212</td>
<td>KU Band Antenna IAE F785</td>
<td>MOD AE17028</td>
<td>Internal Mechanical Provisional Installation Revision 1: add 7 external reinforcement doublers to increase fatigue/damage tolerance figures.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Master Data List Ref AE17028ML Revision 5 dated 25.05.2019. Instructions for Continued Airworthiness Ref. AE17028CA Revision 6 dated 25.06.2018 or later revisions of the above listed documents approved by EASA.</td>
<td>None</td>
<td>25/05/2018</td>
</tr>
<tr>
<td>LEARJET Inc.</td>
<td>REV. 1</td>
<td>FAA ST04091CH</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td></td>
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<td></td>
<td>1. Supplemental Type Certificate FAA ST01432WI-D which is valid through EASA STC 10016518 (previously EASA.A.5.0209) or BombardierService Bulletin 604-34-057 Revision 3 is a prerequisite for this STC</td>
<td>20/07/2018</td>
</tr>
<tr>
<td>IPR CONVERSIONS (SWITZERLAND) LTD</td>
<td>REV. 1</td>
<td>ATR 72-212</td>
<td>Stand Alone Large Cargo Door (LCD)</td>
<td>Transfer of EASA STC 10062773 from Alenia North America - Canada to IPRConversions Switzerland</td>
<td></td>
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<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST0805W</td>
<td>SEE FAA AML</td>
<td>Modification of Rockwell Collins TCAS II System Installation with DCpowered TCAS Transmitter Receiver Processor</td>
<td></td>
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<tr>
<td>NAECAER AVIATION GROUP S.p.A.</td>
<td>REV. 1</td>
<td>A97139 AB139</td>
<td>AW139</td>
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<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC SA1034WI-D</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Modification of a Collins ALT-50A or ALT-65B Radio Altimeter systeminstallation to a Collins ALT-1000 or ALT-4000 RadioAltimeter systeminstallation.</td>
<td></td>
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<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC SA1034WI-D</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Modification of a Collins ALT-50A or ALT-65B Radio Altimeter systeminstallation to a Collins ALT-1000 or ALT-4000 RadioAltimeter systeminstallation.</td>
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**List of EASA Supplemental Type Certificates**
<table>
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<tr>
<th>STC Holder</th>
<th>STC Number</th>
<th>Rev. Level</th>
<th>Foreign/NAAC STC ref.</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>AERODISA S.L.</td>
<td>ATR 72-212, ATR 72-212 A, ATR 72-212, ATR 72-211, ATR 72-102, ATR 72-201, ATR 42-300, ATR 42-400, ATR 42-200, ATR 42-300</td>
<td>REV. 4</td>
<td></td>
<td>Conversion from passenger to allcargo configuration (Class II compartment)</td>
<td>Rev. 2: Update of the Supplement to AFM - Update of the Supplement to AMM, and Adding option to remove the electronic cockpit door locking system. Rev. 3: Compliance with regulations 28...</td>
<td>The EASA Certification Basis for the original product as amended by the following additional alternative airworthiness requirements: The following paragraph(s) at a later amendment: 25.1, 25.23, 25.25, 25.27, 25.29, 25.301, 25.303, 25.305, 25.307, 25.371, 25.471, 25.477, 25.479, 25.489, 25.491, 25.493...</td>
<td>For ATR 72 Master Data List AED_01_2003_00 ED/Rev.0 dated 03/02/2023</td>
<td>Supplement to AFM AED_01_2003_(7Supplements to MPD AED_01_2003_11-F)</td>
<td>16/02/2023</td>
<td>Active</td>
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<tr>
<td>RESURGENT AVIATION SOLUTIONS, LLC</td>
<td>SA04166CH-A</td>
<td></td>
<td></td>
<td>Removal of isophoric coating from the area surrounding static parts on EA500 aircraft.</td>
<td></td>
<td></td>
<td></td>
<td>1. Installation of this approval does not alter the EA500 certification for flight into known icing 2. Prior to installation of this design change it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no...</td>
<td>03/08/2017</td>
<td>Active</td>
</tr>
<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SR06919RC-D</td>
<td></td>
<td></td>
<td>Installation of a collective activated hourmeter.</td>
<td></td>
<td></td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/08/2017</td>
<td>Active</td>
</tr>
<tr>
<td>DASSAULT FALCON SERVICE</td>
<td>FALCON 2000, FALCON 2000EX</td>
<td>REV. 1</td>
<td></td>
<td>Installation of collective activated hourmeter.</td>
<td></td>
<td></td>
<td></td>
<td>Installation of Indurum Satcom FANS 1/A-capable and/or Immant L Band/Fans inch and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/07/2019</td>
<td>Active</td>
</tr>
<tr>
<td>SABENA TECHNICS BGC</td>
<td>737-800-700-800-900-900ER</td>
<td>REV. 1</td>
<td></td>
<td>USB charging plug into cockpit. Installation in cockpit, on both Captain and F/O sides, of one 28/5VDC converter dual USB charging plug into cockpit.</td>
<td></td>
<td></td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/06/2017</td>
<td>Active</td>
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<tr>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td></td>
<td>Infrared Flood Lamp.</td>
<td></td>
<td></td>
<td></td>
<td>Limited to Helionix cockpit variant. Prior to installation of this design change it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/08/2017</td>
<td>Active</td>
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<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>OUTERLINK CORPORATION</td>
<td>FAAA FTD</td>
<td>SR00368BO</td>
<td>EC135 T2, EC135 T3, EC135 P3, EC135 T1, EC135 P2, EC135 P2*</td>
<td>Installation of an Outerlink SATCOM Communications and Flight Data Monitoring System Kit 135-SAT-100. The Certification Basis for the original product as amendability to flight data monitoring. Additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS27.610 at Amptd. 47. The requirements for environmental protection and the associated certified noise and/or emissions levels.</td>
<td>1. Outerlink Corporation's Master Data Drawing Lists (MDL) Document Number: MDL 135-SAT-100. Revision IR, dated 18 February 2016, or later FAA approved revisions of these documents. 2. Operation must be in accordance with Outerlink Corporation’s Rotorcraft Flight Manual Supplement (RFMS) Document Number FMS145SAT001. Revision IR, dated 24 October 2016. 3. Prior to installation of this design change it must be...</td>
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<td>16/10/2017</td>
<td>Active</td>
<td>10062841</td>
<td></td>
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<tr>
<td>OUTERLINK CORPORATION</td>
<td>FAA STC</td>
<td>SR00374BO</td>
<td>MBB-BK117 C-2</td>
<td>Installation of a Satcom and Flight Data Monitoring System Kit 145-SAT-100. The Certification Basis for the original product as amendability to flight data monitoring. Additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS23.01, CS 25.303, CS 25.306, CS 25.307, CS 25.301, CS 25.307, CS 25.341, CS 25.471, CS 25.601, CS 25.603, CS 25.607.</td>
<td>1. Outerlink Corporation's Master Drawing List Document Number: MDL 145-SAT-100, Revision A, dated 24 February 2016, or later FAA approved revision. 2. Operation must be in accordance with Outerlink Corporation’s Rotorcraft Flight Manual Supplement (RFMS) Document Number FMS145SAT001. Revision IR, dated 24 October 2016. 3. Prior to installation of this design change it must be...</td>
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<td>16/10/2017</td>
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<tr>
<td>OUTERLINK CORPORATION</td>
<td>FAA STC</td>
<td>SR00362BO</td>
<td>S-92A</td>
<td>Installation of a Satcom Communications and Flight Data Monitoring System Kit 590-SAT-100. The Certification Basis for the original product as amendability to flight data monitoring. Additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS23.01, CS 25.303, CS 25.306, CS 25.307, CS 25.301, CS 25.307, CS 25.341, CS 25.471, CS 25.601, CS 25.603, CS 25.607.</td>
<td>1. Outerlink Corporation's Master Drawing List (MDL), Document number: MDL 590-SAT-100, Revision B, dated 2 November 2015, or later FAA approved revision. 2. Operation must be in accordance with Outerlink Corporation’s Rotorcraft Flight Manual Supplement (RFMS) Document Number FMS590-SAT-001, Revision B, dated 17 December 2015. 3. Prior to installation of this design change it must be...</td>
<td></td>
<td>11/08/2017</td>
<td>Active</td>
<td>10062857</td>
<td></td>
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<tr>
<td>IPR CONVERSIONS (SWITZERLAND) AG</td>
<td>REV. 1</td>
<td>ATR 72-211, ATR 72-212, ATR 72-201, ATR 72-202, ATR 42-300, ATR 42-320</td>
<td>S-92A</td>
<td>Installation of an auxiliary battery on ATR 42/72 for Reason for Revision: Transfer of STC 100428B8 from ASI INNOVATION to IPRCONVERSIONS (SWITZERLAND) AG. The Certification Basis for the original product as amendability to flight data monitoring. Additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS23.01, CS 25.303, CS 25.306, CS 25.307, CS 25.301, CS 25.307, CS 25.341, CS 25.471, CS 25.601, CS 25.603, CS 25.607.</td>
<td>Master Drawing List ASI-MDL-0463-02 Rev 02 dd. 02.08.2017, ATR Flight Manual Supplement (RFMS) Document Number: ATR-0463, Revision 01 dd. 30.08.2017, ATR Flight Manual Supplement ASI-MPS-0463-02 Issue 02 dd. 01.08.2017. 2. Operation must be in accordance with the above listed documents approved by EASA.</td>
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<td>16/11/2017</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>A321-251N/-261N/-271N A321-2311/-232 A321-2111/-2111-2111+ A321-1111-1112-1113-1114-1115</td>
<td>A320-231N/-232 A320-2111/-2121-2141/215-216 A321-1111-1112-1113-1114-1115</td>
<td>MOD AE17111 - A319-A320-A321 Generic WIFE Systems Activation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List ASI-MDL-0463-02 Rev 02 dd. 02.08.2017, ATR Flight Manual Supplement AE17111AFM Rev 0 dated 06/08/2017. 2. Operation must be in accordance with the above listed documents approved by EASA.</td>
<td></td>
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**List of EASA Supplemental Type Certificates**

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**An agency of the European Union**

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**03/07/2024**
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<td>10062869</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST01309-W-D</td>
<td>SEE FAA AML</td>
<td>Modification of a Collins ALT-550 or ALT-550 Radio Altimeter installation to a Collins ALT-1000 or ALT-4000 Radio Altimeter system installation. Rev. 1: implementation of modified Collins ALT-1000 or 4000 radio altimeters design (new PRN) to allow unattended operations in airworthiness with 5G C-B...)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Collins Drawing Index as listed on AML No. ST01309-W-D RCAA-0728, Revision -, dated 8 December 2023+ (rev) of the above listed document(s) is approved/revised on behalf of EASA in accordance with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>1/02/2024</td>
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<tr>
<td>10062870</td>
<td>ARMSTRONG AEROSPACE Inc/dba ASTRONICS ARMSTRONG AEROSPACE</td>
<td>FAA STC ST01203 CH</td>
<td></td>
<td>757-200</td>
<td>EASA validation of FAA STC ST01203 CH. Installation of a Zodiac RA/4Vh Flt Entertainment System in accordance with StandardAero Master Data List. Doc. No. DC104-9752C-00, Revision (B) dated 30th May 2017 otherwise approved revisions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>The STC installation must be performed in accordance with StandardAero Master Data List, Doc. No. DC104-9752C-00, Revision (B) dated 30th May 2017, and the Airplane Flight Manual Supplement document number DC303-9752C-00, Rev (A) dated 26th May 2017 is a required part of this modification, and the aircraft.</td>
<td>1/02/2017</td>
<td>Active</td>
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<td>10062872</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>REV. 8</td>
<td>FAA STC SA1849W</td>
<td>SEE FAA APPROVED MODELLIST</td>
<td>Installation of Garmin GFC 600 Automatic Flight Control System Rev. 7 amends the FAA Approved Model List with PRM-31-5G(49) and add the MMEL for King Air 200 series Rev. 8 adds TextronAviation Inc. F90 to the AML.</td>
<td>The Type and OISD Certification Basis for the original product amends the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment. CII MMEL is 3; the technical requirements indicated on FAA STC SA1849W amended on date 11th February 2022. The...</td>
<td>- Garmin Master Drawing List (MDL) (055-11065-01) Revision 1, dated 14 July 2017- Garmin GFC 600 Automatic Flight Control System Part 23 AML STC Maintenance Manual (includes instructions for Continued Airworthiness) 190-01938-00, Revision 1 dated 14 July 2017- EASA MMEL report 190-01937-36, MMEL rev. 1...</td>
<td>The STC is valid for aircraft types and models approved by EASA andlifestyled on the FAA AML STC SA1849WI dated 8 December 2023. Prior to installation of this change repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will...</td>
<td>2/11/2023</td>
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<td>10062878</td>
<td>ADAC LUFTFAHRT TECHNIK GmbH</td>
<td></td>
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<td>MBB-BK 117 D-2</td>
<td>Installation of a SpectraLight SX-7 Searchlight onto the multifunctional of an MBB-BK 117 D-2</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement, H145-04109-025, MFS-00, Rev 0/ or later revisions of the above listed documents approved by EASA.-Master Document List, H145-04109-001, UND-0, Rev. 0, dated 22/06/2017- Instructions for Continuing Airworthiness, H145-04109-023, MMS-00, Rev. 0, dated 19/06/2017.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>4/09/2023</td>
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<td>10062883</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST04086 AT-D</td>
<td>GIV</td>
<td>Installation of FANS-1/A+ functionality/this modification is installing a Future Air Navigation System/FANS-1/A+ including the ATS Facilities Notification (AFN), Controller-Pilot Data Link Communication (CPDLC), Traffic and Weather Automatic Dependent Surveillance (ADS-C) coupled with Data Link capability.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unachieved and remain applicable to this certificate/approval.</td>
<td>- Index List, ref : CE41 4968001 revision B, dated 28 March 2016- Airworthiness (including ESAC Document), ref : CE41 4968001, dated 28 March 2016- Instruction for Continued Airworthiness, ref : CE41 4968001 RevisionA, dated 11 January 2016 for later revisions of the above listed documents approved by EASA inaccordance...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>4/09/2017</td>
<td>Active</td>
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<th>Date of Issue</th>
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<tr>
<td>ELDEC LLC</td>
<td></td>
<td>F.A. 1528 STO25348E</td>
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<td>ERJ 190-200 STD</td>
<td>Installation and maintenance of a Crane Aerospace &amp; Electronics TirePressure Sensor Assembly.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise previously installed design change and/or repair will not introduce a noise effect upon the airworthiness of the product.</td>
<td>15/06/2017</td>
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<td>DOA1 LIMITED</td>
<td>REV. 1</td>
<td></td>
<td>A319-115</td>
<td>Introduction of VIP Cabin Reconfiguration Rev. 1: Transfer of STC holder role to DOA1 Ltd.</td>
<td>The Certification Basis The Type and OSD Certification Bases for the original product as amended by the following additional or alternative environmental requirements: Special Condition (s) 1-1: Medical Evacuation/2 Isolated Compartment F-02 Security Protection of Aircraft Systems and Network/</td>
<td></td>
<td>Restricted to AS 1485.Any changes to the WLAN or interior layout approved under this design change will require a re-examination to determine if the compliance demonstration and/or certification bases are still valid.Airworthiness approval of the Airborne Wireless Local Area Network (WLAN) does no...</td>
<td>24/10/2023</td>
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<td>10062886</td>
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<tr>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>EC135 T3</td>
<td>Installation of Multi-Sensor Turret and Recorder.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement AHU/K135-FMS-105, Issue 9 or later revisions of the above listed documents approved by EASA. - STC TWENTY ONE MASTER DOCUMENT LIST No. S21-MDL-1381 ISSUE 1 -</td>
<td>Limited to helicopter with Helionix cockpit variant. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a noise effect upon the airworthiness of the product.</td>
<td>15/06/2017</td>
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<td>AVIANOR INC</td>
<td>TCCA O-LSM17-073D</td>
<td>A330-343</td>
<td>Reconfiguration forward cabin(+/-) install seats and approve new layout of passenger accommodation (LOPA).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AVIANOR MASTER DOCUMENT LIST/1 MDL1618201 REV. NC and MDL16182701 Rev NC</td>
<td>Restricted to MSN 1058.Aerospace minor modification no. MOD 4996-03-A and 365 Aerospace minor modification no. MOD-0305 are prerequisites for this STC.</td>
<td>15/06/2017</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A340-642</td>
<td>Cabin Reconfiguration.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative environmental requirements for: Equivalent Safety Finding(s): CRI D-05 Side facing seats, The following paragraph(s) at a later amendment: 25.603, 25.788 &amp; 25.1447, Appendix B at CS-25 Amrd 19 Th...</td>
<td>Master Data List MDL- FK03-2500 REV. B dated 15-Aug-2017.Airplane Flight Manual Supplement AFMS-FK03-2500 Rev. A dated08-Aug-2017. Instructions for Continued Airworthiness ICA-FK03-2500 Rev. A dated05-Jul-2017 for later revisions of the above listed documents approved by EASA.</td>
<td>Per Airplane Flight Manual Supplement AFMS- FK03-2500.EASA STC 10042295 Rev 2 “Eminence” VVIP interior completion is an irreparable installation.Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed...</td>
<td>15/06/2017</td>
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<td>10062890</td>
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<td>ESG ELEKTRONIKSYSTEEMEN UND LOGISTIK-GmbH</td>
<td>EC135 T3</td>
<td>Installation of Tactical Copilot Display</td>
<td>The Certification Basis (CB) of the EC135/635 P3(CPDS)/53(CPDS) is applied to all models listed on this certificate. Refer to TCDS/EASA/R.009, Issue 11, or later revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are...</td>
<td>Flight Manual Supplement &quot;Installation of Tactical Copilot Display&quot; for Airbus Helicopters EC135.Doc.-ID 00030-DO-MZ-LCD-00, Iss...</td>
<td>Not applicable to EC135/EC635 models with Helionix cockpit instrumentation. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a noise effect upon the airworthiness of the product.</td>
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<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>A330-321/341-342/343</td>
<td>MOD CJ2087 INSTALL OF A330 2KU ANTENNA Revision 1 of the STC 10062892 introduces additional pre-modification configuration conditions for installation of a GoGo 2KU antenna on an airplane that requires relocation of Emergency Locus Transmitter (ELT) antennae(s) in accordance with Airbus major modification...</td>
<td>Certification Summary Reports ref. ACJC-AJ-1403-001; Aircraft Approval Sheet ref. MAS CJ2087 Issue 1 (Major Change/Maintenance Planning Document ref. MDP-A330 MOD CJ2087 Revision 05) later versions of the above listed documents(s) approved/ acceptable/ replaced.</td>
<td>Modification of an airplane in accordance with the Revision 1 of STC10062892 is allowed, provided that prior to GoGo antenna installation the airplane is modified in accordance with Airbus SB A330-25-3705 (Airbus MOD 207956), or Airbus SB A330-25-3703 (Airbus MOD 207959), applicable to airplane m...</td>
<td>22/10/2019</td>
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<td>326 SUPPORT SERVICES GmbH</td>
<td>REV. 2</td>
<td></td>
<td>GULFSTREAM 200/ GALAXY</td>
<td>A350 B-OUT on RC Pro Line 4 (G-200)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Modification(s): CRI F-01, Continuity Requirements for ADS-B function. - The following paragraph(s) at a later amendment: CS-ACNS, Initial Issue: CS-ACNS (Airbus Structure), CS-ACNS (Aircraft)-001 (CS-ACNS) (Aircraft),CS-ACNS (Aircraft)-002</td>
<td>Master Data List (MDL). Document No. MDDL1/32/01, Rev. 3; AFM Supplement, Document No. AFMS016C342001; Rev. 8; MMEEL Supplement, Document No. MMELS016C342001, Rev. 3; Aircraft later revisions of the above listed documents approved by EASA.</td>
<td>Limited to S/N 062 and up Prior to installation of this design change it must be determined that the interaction relationship between this design change and any other previously installed design change and/or repair will introduce route adverse effect on the airworthiness of the product.</td>
<td>23/10/2019</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>REV. 4</td>
<td>A340-643/ A340-642</td>
<td>Tailwind 550 SAT TV antenna installation</td>
<td>The Certification Basis (CS) for the original product remains applicable to this certificate/approval Equivalent Safety Finding(s): CRI B-01 Vibration / Buffeting compliance criteria with additional criteria for large external antenna installation. The requirements for environmental protection and the associated certified noise...</td>
<td>- Master Data List MDD: A34021-1500 Rev. B dtd. 17/08/2017. - Airplane Flight Manual Supplement AFMS: A34021-1500 Rev. A dtd. 08/08/2017. - Instructions for Continued Airworthiness ICA-A34021-1500 Rev. C dtd. 17/08/2017 later revisions of the above listed documents approved by EASA.</td>
<td>Master Data List MDD: STR001003 Revision 2, Aircraft Flight Manual Supplement AFMS: STR001003; Revision 3, dtd. 10.10.2022 Instruction(s) for Continued Airworthiness ICA- STR001003, Revision 3 dated 10.10.2022 later revisions of the above listed documents(s) approved/ acceptable/ replaced.</td>
<td>Per - Airplane Flight Manual Supplement AFMS: A34021-1500-000 - Instructions for Continued Airworthiness ICA-A34021-1500/Prior to installation of this design change it must be determined that the interaction relationship between this design change and any other previously installed design change and/or repair will introduce adverse effect on the airworthiness of the product.</td>
<td>17/08/2017</td>
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<td>STRAERO S.A. (INSTITUTE FOR TESTING AND CALCULATIONS STRUCTURES AEROSPACE)</td>
<td>REV. 4</td>
<td>BNA-27</td>
<td>Installation of surveillance and atmospheric measurement equipment incl. cabin refreshment as defined in STRAERO MDD, STR001003, Rev. 0, dated 10 AUG 2017 Revision 1 Installation of surveillance and atmospheric measurement equipment incl. cabin refreshment as defined in STRAERO MDD, STR001003, Rev. 0, dated 10 AUG 2017 Revision 1</td>
<td>Installation of surveillance and atmospheric measurement equipment incl. cabin refreshment as defined in STRAERO MDD, STR001003, Rev. 0, dated 10 AUG 2017 Revision 1</td>
<td>- Master Data List MDD: STR001003 Revision 2.</td>
<td>Master Data List MDD: STR001003 Revision 2.</td>
<td>This STC is limited to BNA-27, S/N 804, 808 and 824. Prior to installation of this change/repair it must be determined that the interaction relationship between this change/repair and anything previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/09/2023</td>
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<td>SABENA TECHNICS BOD</td>
<td>REV. 7</td>
<td>A350-941</td>
<td>Seats and floor covering installation The modification consists in a 411 economy/ premium economy seats installation, including the seat power system and in-Fight/Entertainment system connections, and the installations of the seat footrest, seat back, cover, cabin placards, and floor path/lining.</td>
<td>Seats and floor covering installation The modification consists in a 411 economy/ premium economy seats installation, including the seat power system and in-Fight/Entertainment system connections, and the installations of the seat footrest, seat back, cover, cabin placards, and floor path/lining.</td>
<td>- Master Data List: ref MDL-5-25-4102 Issue K, dated 12 October 2022 - Instruction for Continued Airworthiness, ref: ICA- V0003-25-009 Issue 3, dated 5th October 2022 - Weight and Balance Manual Supplement, ref: WBMS-0003-25-039 Issue A, dated August 2017.</td>
<td>Prior to installation of this change/repair it must be determined that the interaction relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this change/repair it must be determined that the interaction relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10062949</td>
<td>GOODRICH LIGHTING SYSTEMS GmbH &amp; Co. KG</td>
<td>REV. 2</td>
<td>A321-271N/-272N</td>
<td>Installation of LED Navigation Light</td>
<td>Rev. 1: updates the ICAO and the installation documentation with additional existing wing navigation light patterns which can be substitutable part of the STC. Rev. 2: minor updates of build standard of LED wing navigation light based on editorial changes. Revision of tech.</td>
<td>Modification Summary Sheet and Declaration of Compliance for Design Changes</td>
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<td>15/03/2024</td>
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<td>10062960</td>
<td>ALPINE AEROTECH LP</td>
<td>REV. 1</td>
<td>TCCA STC SH00-2</td>
<td>Firewall Web Installation (Alpine Aerotech Ltd. P/ N AAL-061-903-001).</td>
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<td>15/06/2017</td>
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<td>10062985</td>
<td>FOKKER SERVICES B.V.</td>
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<td>DHC-8-401 DHC-8-402</td>
<td>ADS-B out on the Dash 8-400 series Installation of the NXT 700 instead of the MST-67A Transponder.</td>
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<td>31/08/2017</td>
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<td>AIRBUS S.A.S.</td>
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<td>A330-243</td>
<td>MOD CJ2083 : Installation of Zodiac KA BAND Antenna on A330.</td>
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<td>25/08/2017</td>
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<td>10063006</td>
<td>AIRCRAFT DEVELOPMENT AND SYSTEMS ENGINEERING (ADSE) B.V.</td>
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<td>737-700</td>
<td>Certification of the Boeing 737-700 as PED tolerant</td>
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<td>28/08/2017</td>
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<td>Associated Technical Dissemination</td>
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<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10063008</td>
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<td>Upgrade of existing cabin lighting to the STC Aerospace iLivMoody® LED Cabin Lighting system.</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: the following paragraph(s) at a later amendment. CS25.1353 at CS25 amendment 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List LM377-MDL-01 rev B dated July 20th 2015.Configuration Matrix LM377-CM-01 rev A dated July 21st 2015 later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004A/04C or (subsequent revisions of this decision) and/or the Technical Implementation.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/08/2017</td>
<td>Active</td>
<td>10063005</td>
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10063015 ALPINE AEROTECH LP REV 1 TCCA STC SH50-36 212/412/412EP Cowling Baffle Screen Installation in accordance with Master Document List (MDL) AAL-232-024-001. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | Master Document List (MDL) Document No. AAL-232-024-001 Revision NL, dated May 27, 2005, or later TCCA approved revision Cowling Baffle Screen Installation Instructions Document No. AAL-222-025-001, Revision NL, dated May 27, 2005, or later TCCA approved revision Cowling Baffle Screen Maintenance Manual... | Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 15/09/2017 | Active | 10063015 |

10063016 JET AVIATION AG MYSTERE FALCON 900 IMP/900 FALCON 900EX (PF00E) (550) 560 560L Display Unit replacement by EID CRT Display Units Honeywell DU-870 model DU-875. Display Unit replacement by EID LCD Display Units Honeywell DU-875. Installed: 1 or up to 5 EFIS and EID CRT Display Units Honeywell DU-875. Removed: 1 or up to 5 EFIS and EID LCD Display Units Honeywell DU-875. | Approved Model List: Doc. No. JBSL-M-00631 AML-01, Issue 01: Instructions for continued airworthiness: Doc. No. JBSL-M-00631 ICA-01, Issue 01: later revisions of the above listed documents approved by EASA Master Document List: Doc. No. JBSL-M-00631 MDL-01, Issue 02: Declaration of Compliance: Doc. No... | Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 29/08/2017 | Active | 10063016 |

10063017 ALPINE AEROTECH LP TCCA STC SH50-30 212/412/412EP Oil Cooler Support Bracket, Kit. | The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: FAR 29 paragraph 395.109(a) and 395.109(b) at FAR 29 amendment 25 at FAR 29 amendment 30. | TCCA Airworthiness Limitations section of Oil cooler support bracket, kit. Maintenance Manual Supplement Document No. AAL-214-015-101, Rev. B dated October 27, 2011, or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU and Canada... | Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 15/09/2017 | Active | 10063017 |

10063018 ALPINE AEROTECH LP TCCA STC SH50-31 212/412/412EP Induction Baffle Screen Installation in accordance with Master Document List (MDL) No. AAL-253-024-001. | Master Document List (MDL) Document No. AAL-253-024-001, Revision NL, dated May 13, 2005, or later TCCA approved revision Induction Baffle Screen Installation Instructions Document No. AAL-253-025-001, Rev. NL, dated May 13, 2005, or later TCCA approved revision Induction Baffle Screen Maintenance Manual Supplement Document No. AAL-253-025-101, Rev.d... | Prior to installation of this design change it must be determined that the interaction between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 15/09/2017 | Active | 10063018 |
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<th>Ref.</th>
<th>STC Holder</th>
<th>STC Type</th>
<th>Rev. Level</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
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<tr>
<td>10063031</td>
<td>ESAO ELEKTORONIKSYSTEMLANDLOGISTIK-GMBH</td>
<td>REV. 3</td>
<td>EC635 T3, EC635 P3, EC635 T1, EC635 T2, EC635 T2, EC635 P2, EC635 P2, EC635 P2, EC635 P2, EC635 P2</td>
<td>Flight Manual Supplement for Installation of EuroNav 7 Digital MovingMap System, 00036-DO-M2-FMS-00, Issue 01 dated 18 August 2017</td>
<td>This change is limited to helicopters which are equipped with CPDLS and dual Monitor FCS. This change excludes helicopters which are equipped with the HELIONIXcockpit. Prior to installation of this design change it must be determined that there is no interference relationship between this design change and any other...</td>
<td>19/02/2017</td>
<td>Active</td>
<td>10063031</td>
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<td>10063040</td>
<td>AEROLITE AG</td>
<td>REV. 1</td>
<td>AW169</td>
<td>- Flight Manual Supplement FMS-169-023000-501-01, A dated 29/03/2019; Master Document List MDL-17-001, Rev. C, dated 28/04/2019; Operations Manual OPM-169-023000-501-A, issue 28/03/2019 for later revisions of the above listed documents approved/ accepted/published the EASA system</td>
<td>Pre-requisites for installation of PIN 169-023000-501 are given in OPM-169-023000-501-A Rev. 2. For any installation onto the Cargo Platform a separate design approval is needed (not scope of this STC). Prior to installation of this change/repair it must be determined that...</td>
<td>06/05/2019</td>
<td>Active</td>
<td>10063040</td>
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<td>10063069</td>
<td>GAMA AVIATION (ENGINEERING) LIMITED</td>
<td>REV. 1</td>
<td>MODELS B200 AND B300</td>
<td>- Installation in accordance with GAMA Document No. MDL-1451, Issue 3 dated 14 December 2017; Master Data List (MDL) for Honeywell ASP-200SatCom Installation. Basestar Corporation King Air B200 / B300 series: Operate in accordance with GAMA Flight Manual Supplement reAFCSDS-1451, Issue 1, date...</td>
<td>Prior to installation of this design change it must be determined that there is no interference relationship between this design change and any other previously installed design change and/or repair will introduce aero dynamic effect upon the airworthiness of the product.</td>
<td>14/12/2017</td>
<td>Active</td>
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<td>EASA STC no.</td>
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<td>10063071</td>
<td>LAZ-TEC, LLC</td>
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<td>FAA STC SR00626DE</td>
<td>429</td>
<td>Installation of Baggage Parcel.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Lao Tzei Design Ltd., Report No. LT-1023, Revision A, dated 16 February 2017, FAA approved 3 March 2017, or later FAA approved revisions. 2. Flight Manual Supplement, Lao-Tzei, LLC Document No. LT-15228, Notice revision, dated 24 February 2017, or later FAA approved revisions. 3. Instructions for Continuation.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse relationship between this design change and any other previously installed design change and/or repair which may introduce nosecone effect upon the aerodynamics of the product.</td>
</tr>
<tr>
<td>10063084</td>
<td>AIRBUS DEFENCE AND SPACE S.A.</td>
<td>REV. 6</td>
<td>A330-243</td>
<td></td>
<td>Conversion of A330-200 Series A/C to A330 Multi Role Transport Tanker(MRTT) GOS (Government of Singapore).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative environmental requirements. The following paragraph(s) at a later amendment: CS 25.0522 Initial Issue for affected areas by this STC/CS 25.0583 Initial Issue for affected areas by change DC01085-E...</td>
<td>- CONVERSION DEFINITION A330 MRTT GOS STC document, Ref. TAE-FG-DT-160003 Issue A. - Aircraft Flight Manual Supplement “EASA APPROVED MODEL A-330 MRTT GOS/ARPLANE FLIGHT MANUAL” document, Ref. TAE-FG- DT-15001, Issue as referenced in STC 10066941 AML TEA-FA-AIR-160001C - Temporary Revision No. 5, Issue 1...</td>
<td>This STC only covers the installation of external aerodynamic devices/underramer pods, boom, receptacle for refueling, antennas, fairings in case they are equipped with the associated structural modifications, the modification of some basic aircraft systems, and the installation of some inoperable equipment.</td>
</tr>
<tr>
<td>10063088</td>
<td>PRATT &amp; WHITNEY ENGINE SERVICES, IN</td>
<td>REV. 1</td>
<td>FAA STC SA122050C</td>
<td>208, 208B</td>
<td>Installation of a Pratt &amp; Whitney Flight data acquisition, storage, and transmission system (FAST) Rev. 1; changes the “Limitations and Conditions” section to align with amended FAA STC No. SA122050C, dated 23 March 2023.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Data List for Cessna 208 Series, document no. GHT- T-100-7 Revision B, dated 1 May 2017 (&lt;&gt;); NEW-PAGE2. FAST Instructions for Continued Airworthiness for Cessna 208 Series aircraft, document no. GHT-T-260-B Revision B, dated 20 April 2017 (&lt;&gt;); 2. Flight Manual Supplement for Cessna 208 Series...</td>
<td>Prior to installation of this design change it must be determined that the design change and any other previously installed design change and/or repair which may introduce nosecone effect upon the aerodynamics of the product. This STC is applicable to Textron Aviation, Inc...</td>
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<tr>
<td>10063096</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>A321-111/131/211/-231/A320-211/214/216-230-251N A319-112/114-132</td>
<td></td>
<td>Removal of K-airband Satcom Removal of EASA STC no. 10068540. Extension to A320-251N within the list of the eligible models.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any improvements.</td>
<td>- Compliance Checklist: CRL23-5776M Rev 3 dated 22 Jan 2020. Instructions for Continued Airworthiness: A-00-17/84-ICA Issue 02 RevA dated 05 Dec 2019. Supplemental Type Design Document: STD202-17-914M Issue 02 Rev A dated 16 Jan 2020. Later revisions of the above listed document(s) approved/accepted...</td>
<td>Airworthiness Limitation Section within A-00-17/84-ICA Prior to installation of this change/repair it must be determined that there is no adverse relationship between this change/repair and any other previously installed change and/or repair which may introduce nosecone effect upon the aerodynamics of the product...</td>
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<td>10063143</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>EC135 T3</td>
<td>EC135T3H NVIS Certification for NVG Operations – AMLA/135/3101.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement AMLA/135-FMS-109 Revision 0: NVIS Certification for NVG Operations(&lt;&gt;) or later revisions of the above listed documents approved by EASA. - Design Compliance Report DCR/1353101-01 Issue A, dated 15 August 2017. - Declaration of Compliance DOC/ 1353101-01, Issue A, d...</td>
<td>Limited to helicopters with Helionix cockpit variant. Any deviation from the cockpit or cabin configuration specified in this STC Type Design may affect the compatibility of the NVIS system with the Helionix/VisionMerging System (NVIS) and may require a re-evaluation for NVIS compatibility. Once the aircraft is modified...</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
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<td>10063149</td>
<td>AVONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>525, 525A</td>
<td></td>
<td>Installation of Standby Instrument L-3 ES-2000.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: C3-23, Amendment 4: (not required according to 21A, 101/29, 291, 301, 325, 307, 337, 561, 601, 603, 605, 607, 609, 611, 613...</td>
<td>Flight Manual Supplement ASR-2017-007-FMS-05-00 Amdt. 00&lt;4&lt;1&lt;8&lt;0&lt;9 later revisions of the above listed documents approved by EASA.- Certification Programme, Compliance Report, Compliance Statement and Declaration of Compliance ASR-2017-007-CEP-02-00, Amdt. 00 - Instructions for Installation ASR-201...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness or/and environmental protection.</td>
<td>14/09/2017</td>
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<tr>
<td>10063152</td>
<td>ANJOU AERONAUTIQUE S.R.L.</td>
<td>737-800</td>
<td></td>
<td>Replacement of seatbelts on 156 aircraft seat.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>02-100074-MDL-001, Rev. 04 dated 16 August 2017 “Master Data List” and/or later revisions of the above listed documents approved by EASA.</td>
<td>The STC approval is limited to Group 1 Potable seatbelts installed on the following seat models: Recaro 3410 Seats. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and...</td>
<td>14/09/2017</td>
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<td>10063155</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>ISAT SYSTEM AND ANTENNA INSTALLATIONInstallation of SkytackSatCom systems ISAT-100 and ISAT-200A (only used them to be installed in the aircraft at the same time) in cockpit avionics bay, and related GPS/Inertial Antenna installation on top offset slat</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s): CRI-F-01 “Liithium battery installations: The requirements for environmental protection and the associated certified noise and/or emissions levels of th...</td>
<td>-S4A Modification Description, ref. doc. 60434-DES01, Issue 1, dated 13/January 2016-S4A Certification Plan, ref. doc. 60434-CP01, Issue 5, dated 31/August 2017- S4A Flight Manual Supplement, ref. doc. 60434-HDK01, Issue 1, dated 22/February 2017-S4A Maintenance Manual Supplement, ref. doc. 60434-HDK01, Issue 1, dated 22/February 2017-S4A Maintenance Manual Supplement, ref. doc. 60434-HDK01, Issue 1, dated 22/February 2017-S4A Maintenance Manual Supplement, ref. doc. 60434-HDK01, Issue 1, dated 22/February 2017-S4A Maintenance Manual Supplement, ref. doc. 60434-HDK01, Issue 1, dated 22/February 2017-S4A Maintenance Manual Supplement, ref. doc. 60434-HDK01, Issue 1, dated 22/February 2017-S4A Maintenance Manual Supplement, ref. doc. 60434-HDK01, Issu...</td>
<td>None</td>
<td>15/09/2017</td>
<td>Active</td>
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<td>10063156</td>
<td>AIRBUS S.A.S.</td>
<td>A319-115</td>
<td></td>
<td>Airbus Interiors Services MOD CJ2158 SECURITY GUARD SYSTEM INSTALLATION</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s): CRI-H-01 “Enhanced Airworthiness Programmes for Aeroplanes Systems – EWIS/CA – The requirements for environmental protection and the associated certified fi...</td>
<td>-Modification Approval Sheet: MAS CJ2158 Issue 02, dated 14th September 2017-MPD Installation Approval: MPS-8 MOD CJ2158 dated 1st September 2017 or later revisions of the above listed documents approved by EASA.</td>
<td>This approval is limited to Airbus A319-115/15N-1556</td>
<td>15/09/2017</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>MOTORFLUG BADEN-BADEN GmbH</td>
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<td>MM6-BK117 C-2</td>
<td>Installation of Dual COMNAV/GPS GT750, G733 Mode-S Transponder and R21 Marker as per Mod. 2510/1F160284-01. This STC is not applicable to MM6 BK117 C-2.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CS-ACS Ins. Initial Issue: B. VC(S) (a), D. EHS (a), D. ELS (a), D. ADSB (a) CS-29 Am 04: 29.1316 and 29.1317. Approvals/products and the associated.</td>
<td>25101 F160284-01-0001 Rotarycraft Flight Manual Supplement, R1, dated14.09.2017 for later revisions of the above listed documents approved by EASA 25101.</td>
<td>15/09/2017</td>
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<td>ALPINE AEROTECH LP</td>
<td></td>
<td>TCCA STC SH06-2</td>
<td>212/ 412/ 412EP</td>
<td>Firewall Cover installation in accordance with MDL AAL-237-014-001.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment: FAR 29 up to Amendment 29-47: 29.1, 29.601, 29.603, 29.605, 29.607, 29.609, 29.613, 29.901, 29.1191, 29.159.</td>
<td>Firewall Cover Master Document List, Document No. AAL-237-014-001, Revision A, dated April 17, 2017, or later TCCA approved revisions.</td>
<td>Firewall Cover Installation Instructions, Document No. AAL-227-015-001 Revision NI, dated September 30, 2005, or later TCCA approved revisions.</td>
<td>15/09/2017</td>
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<td>MILLENNIUM CONCEPTS, INC.</td>
<td></td>
<td>FAA STC SA0171W1</td>
<td>KODIAK 100</td>
<td>Executive Style Interior Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>1. Millennium Concepts Master Drawing List, No. 2038-MDL025 Rev. A, dated 10 July 2014. 2. Millennium Concepts Aircraft Flight Manual Supplement No. 2038- AFM651. Rev. A, dated 27 August 2014, or later FAA approved revisions.</td>
<td>For production, on aircraft serial numbers 100-0069, 100-0081, and 100-0084 through 100-0097, STC SA0171W1 must be installed prior to modification. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/09/2017</td>
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<tr>
<td>ANDRA INTERNATIONAL LLC</td>
<td></td>
<td>FAA STC SA0014NH</td>
<td>747-400F 747-200B/-200C/-200F/-30</td>
<td>Installation of Ancra Cargo System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Ancra Drawing List DL 82000, Revision B, dated 17 August 1990, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/09/2017</td>
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<td>WIPARE Inc.</td>
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<td>FAA STC SP04133CH</td>
<td>4HFR34C378</td>
<td>Installation of pitch locks.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Wipaire, Inc. Installation Instructions, Document No. 1008767, Revision C, dated 17 June 2016, or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/09/2017</td>
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<tr>
<td>PPG AEROSPACE</td>
<td>REV 1</td>
<td>FAA STC ST02542A</td>
<td>HAWKER 900XP HAWKER 800XWP</td>
<td>Installation of PPG Aerospace cockpit windshields PNs 196600-01 and 196600-02.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>1. PPG Aerospace Master Data List 196600 at Revision B, dated 12 August 2015, or later FAA approved revision. 2. Instructions for Continued Airworthiness (ICA) Document No. U10-0370, Revision B, dated 11 August 2015 or later FAA accepted revision.</td>
<td>This approval requires the 196600-01 and 196600-02 windshields be maintained according to the following PPG Aerospace component maintenance manual (CMM), 56-10-23, original issue, dated 7 August 2015 for other FAA accepted maintenance data. Prior to installation of this design change it must be determined...</td>
<td>21/09/2017</td>
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<td>Limitations/Conditions</td>
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<td>10063181</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>REV 1</td>
<td>FAA STC STO2540SE</td>
<td>777-300ER, 777-200LR</td>
<td>Interior reconfiguration.</td>
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<td>11/06/2018</td>
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<td>10063197</td>
<td>AVODYNE CORPORATION</td>
<td>FAA STC SAU2013CH</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of Avodyne Corporation Traffic Advisory System (TAS) in accordance with the Master Document (MDL) referenced on the Aircraft Service Bulletin.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td></td>
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<td>22/09/2017</td>
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<tr>
<td>10063200</td>
<td>WIPARE Inc.</td>
<td>FAA SAU414CH</td>
<td>208, 208B</td>
<td>Installation of modified McCauley Propeller (S0A0153CH) on Cessna 208B and 208B (SAU414CH3).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
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<td>20/06/2017</td>
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<td>10063209</td>
<td>WORLDWIDE AVIATION L.L.C.</td>
<td>FAA STC STO1836W</td>
<td>SAAB 2000</td>
<td>Installation of a Reinforced Flight Deck Door.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td></td>
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<td>21/09/2017</td>
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<td>10063212</td>
<td>STENE AVIATION, INC.</td>
<td>FAA STC SAD2513SE</td>
<td>SEE FAA AML</td>
<td>Installation of wing tips with integrated Light Emitting Diode (LED) landing and taxi lights</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td></td>
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<td>21/09/2017</td>
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<td>EASA STC No</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10063214</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>REV. 4</td>
<td>A9189</td>
<td>&quot;Quick-conversion&quot; Air Ambulance Equipment, installing various EMS cabinlayout configurations: Rev. 1 covered minor STC certificate editorial change; Rev. 2 covered the addition of Oxygen System under &quot;Complete system&quot; or &quot;Provisions Only&quot; to the previously approved EMS configurations reinstating one o...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchange and remain applicable to this certificate/approval.</td>
<td>For Configuration AW 189-25-20-101.000-601/01 1-180,1 (EMI) (Equipmrnt max. 1 patient): Master Data List MDL AW 189-25-20-101.000-601, Rev. 0: Certification program CP, A9189-25-20-101.000-60X /70x, Rev. 1: Compliance Report CR, A9189-25-20-101.000-601, Rev. 0: For Configuration AW 189-25-20-101.000...</td>
<td>Installation of oxygen cylinders and/or pressure regulator for the &quot;Oxygen Provisions Only&quot; configuration -604 requires further approval by the Competent Authority. Prior to installation of this design change it must be determined that the international relationship between this design change and any other prev...</td>
<td>02/06/2021</td>
</tr>
<tr>
<td>10063232</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>DORNER 228-201</td>
<td>Installation of Cabin Emergency Lights in accordance with ATL Modification Summary Sheet (ATL/228/M001 Issue 3) dated 06/08/204 after major revision. Right Manual Supplement ATL Supplement No20 Rev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements LBA TCDS 203/13: The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchange and remain applicable to this certificate/approval.</td>
<td>ATL DDP 503 Issue 2 dated 29/11/2004 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>1. Subject to compliance with the provision of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered. Withdrawal otherwise terminated. 2. If transfer of this certificate is requested under Part 21A.116, the STC be released. 3. This STC is approved only for the product...</td>
</tr>
<tr>
<td>10063233</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 2</td>
<td>757-200-300/20PF</td>
<td>737-700-700/800-900/900-9008</td>
<td>Modification ATL/CPT/M138 Issue 1 dated 29/11/2004, Business Class Double Seat Series C227/2055(16),499/Rev. 2: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchange and remain applicable to this certificate/approval...</td>
<td>For Configuration AW 189-25-20-101.000-601/01 1-180,1 (EMI) (Equipmrnt max. 1 patient): Master Data List MDL AW 189-25-20-101.000-601, Rev. 0: Certification program CP, A9189-25-20-101.000-60X /70x, Rev. 1: Compliance Report CR, A9189-25-20-101.000-601, Rev. 0: For Configuration AW 189-25-20-101.000...</td>
<td>1. Subject to compliance with the provision of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawal otherwise terminated. 2. If transfer of this certificate is requested under Part 21A.116, the STC be released. 3. This STC is approved only for the product...</td>
</tr>
<tr>
<td>10063234</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td>757-200-300/20PF</td>
<td>737-700-700/800-900/900-9005</td>
<td>Conversion of Reyward Business Class Double Seat Series C327/075(16), Rev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchange and remain applicable to this certificate/approval...</td>
<td>ATL DDP 503 Issue 2 dated 29/11/2004 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>1. This STC relates only to the modification of the seat structure. Aseparate modification will be needed for a passenger seat layout using these seats and for use off/ compatibility with any seat mounted passengerentertainment equipment. 2. This STC is approved only for the product...</td>
</tr>
<tr>
<td>10063235</td>
<td>SYLT AIR GmbH</td>
<td>LBA STC 1294</td>
<td>P.68 B &quot;VICTOR&quot;, P.68 C</td>
<td>Bubble Window Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchange and remain applicable to this certificate/approval.</td>
<td>B: 45, 170, 212, and P.68 C: 455 only! The approval holder shall fulfill the obligations of Part 21, Paragraph 21.A.109 or 21.A.45(b) (as applicable) Prior to installation of this design change it must be determined that the international relationship between this design change and any other prev...</td>
<td>Aircraft Serial Numbers for P.68 B: 45, 170, 212, and P.68 C: 455 only! The approval holder shall fulfill the obligations of Part 21, Paragraph 21.A.109 or 21.A.45(b) (as applicable) Prior to installation of this design change it must be determined that the international relationship between this design change and any other prev...</td>
<td>27/06/2017</td>
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<tr>
<td>10063239</td>
<td>ROEDER PRECISION GMBH</td>
<td>REV. 1</td>
<td>SA227-DC, SA227-TC, SA227-AT, SA226-TB, SA226-T, SA226-TC, SA227-AC, SA227-8C</td>
<td>Installation of MST 67 Mode-S Transponder.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unchange and remain applicable to this certificate/approval.</td>
<td>AFM Supplement SA226/227 NY-003-08, Rev. 0, or later EASA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the international relationship between this design change and any other prev...</td>
<td>25/09/2017</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC Type</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>10063240</td>
<td>ROEDER PRAEZION GMBH</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>SA227-CC, SA227-CC, SA226-AT, SA227-AT, SA227-CC, SA227-AC, SA227-BC, SA226-T(8)</td>
<td>Installation of a Garmin GTX 330D Mode S Transponder.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and associated aircraft noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Flight Manual Supplement Doc. No. SA226/227-NY-0064-09 or later approved revision. - MM Supplement SA226/227-NY-0064-07 or later approved revision.</td>
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<tr>
<td>10063257</td>
<td>AVIONITEC Ltd.</td>
<td>172P</td>
<td></td>
<td></td>
<td>Installation Garmin GNS 430W/Remov. Bendix/King KLX 135A COM/OPS and Bendix King KA 92 GPS Antenna Installed. Garmin GNS 430W COM/NAV/OPS, Garmin GA 35 WAAS Antenna and Garmin G100B NAV Indicator Compliance to AMC20-28 and CFR F.10 was show for RNAV approaches down to LNAV, LNAV/NAV and LPV Level.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Based on 14 CFR §§ 21.115 and 21.1010(b)(1), and FAA Order 8110.48, the certification basis is per the AML TCDS plus the following Regulations at the latest amendment 25-0...</td>
<td>Flight Manual Supplement: Doc. No. 7687-011, dated 17.07.2017 for later revisions of the above listed documents approved by EASA. Master Document List: Doc. No. 7687-017, dated 17.07.2017.</td>
<td>Limited to S/N 17276660 (HB-CLS). Technical compliance to AMC20-28 and CFR F.10 does not constitute operational approval. It is the operator’s responsibility to gain approval from the applicable airworthiness authority prior to installation of this design change. The requirements for the above listed documents approved by EASA. See FAA AML Supplement 2540-HDK01, Rev. 5, dated 17.07.2017.</td>
</tr>
<tr>
<td>10063260</td>
<td>LIBERTY PARTNERS, INC.</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>SEE FAA AML</td>
<td>ST110715C</td>
<td>Demonstration of aircraft critical equipment and antenna systems/obstacle of radio frequency interference (RFI) from intentional on-ground transmissions from certain Portable Electronic Devices (PEDs)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Based on 14 CFR §§ 21.115 and 21.1010(b)(1), and FAA Order 8110.48, the certification basis is per the AML TCDS plus the following Regulations at the latest amendment 25-0...</td>
<td>1. Master Drawing List L60-DL001, revision H, dated 14 March 2017, or later approved FAA revision. 2. One of the following Airplane Flight Manual Supplements (AFMS): document L60-FM001, L60-FM002, L60-FM003, L60-FM004, L60-FM005, L60-FM006, L60-FM007, L60-FM008, L60-FM009, L60-FM010, L60-FM011, L60-FM012, L60-FM013.</td>
</tr>
<tr>
<td>10063263</td>
<td>SEACRPLANE TECHNOLOGIES, INC.</td>
<td>FAA STC</td>
<td>SEE FAA AML</td>
<td>ST025005E</td>
<td>SAAB SF340A, SAAB 340B</td>
<td>Installation of Seacrplanes Sealed Lead Acid Batteries.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Based on 14 CFR §§ 21.115 and 21.1010(b)(1), and FAA Order 8110.48, the certification basis is per the AML TCDS plus the following Regulations at the latest amendment 25-0...</td>
<td>1. Master Data List, AM0300-DL02-00 Revision A, dated 4 March 2017, or later FAA approved revision. 2. Instructions for Continued Airworthiness AM0300-IC02, Revision A, dated 21 February 2017, or later FAA approved revision. 3. Airplane Flight Manual Supplement, AM0300-FM01-00, Revision -. FAA Approved.</td>
</tr>
<tr>
<td>10063264</td>
<td>SAA SOLUTIONS FOR AVIATION, S.L</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>208B</td>
<td>Video Mast installation in accordance with SAA Service Bulletin 20840-SB001.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Based on 14 CFR §§ 21.115 and 21.1010(b)(1), and FAA Order 8110.48, the certification basis is per the AML TCDS plus the following Regulations at the latest amendment 25-0...</td>
<td>- Flight manual supplement 20840-HDK01, Rev. 5, dated 20 December 2017. - Flight manual supplement 20840-HDK02, Issue 3, dated 20 December 2017.</td>
</tr>
<tr>
<td>10063273</td>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>FAA STC</td>
<td></td>
<td></td>
<td>ST01414AT</td>
<td>Interior Reconfiguration.</td>
<td>The Certification Basis (CB) for the original product as amended by the following additional or alternative airworthiness requirements: Based on 14 CFR §§ 21.115 and 21.1010(b)(1), and FAA Order 8110.48, the certification basis is per the AML TCDS plus the following Regulations at the latest amendment 25-0...</td>
<td>1. TIMCO Aerosystems Master Data List (MDL), A13442001, Revision C, dated 14 July 2014, or later FAA approved revision. 2. TIMCO Aerosystems Instructions for Continued Airworthiness (ICA)/3442R009, Rev. IR, dated 23 May 2014, or later FAA approved revisions.</td>
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<tr>
<td>STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA</td>
<td>STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10063277</td>
<td>AIRBUS S.A.S.</td>
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<td>A318-112</td>
<td>Airbus Interiors Services MOD CJ2154 - AL MAJD A318 MSN1599 Cabin Communication System Upgrade including WLAN</td>
<td>The Certification Basis for the original product as amended/adding additional or alternative airworthiness requirements: Special Condition (s): CRI-F-0304-000 WLAN &amp; Interfacing T-PEDS issue 1. The requirements for environmental protection and the associated certified noise and/or emissions level.</td>
<td>-Modification Approval Sheet MAS CJ2154 issue 01, dated 29th September 2017. AFM - Supplement, ref. doc. FMS MOD CJ2154, dated 1st July 2017 or later revisions of the above listed documents approved by EASA.</td>
</tr>
<tr>
<td>10063283</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>777-300ER</td>
<td></td>
<td>Gogo 2KU - Structural Provisions Rev. 1: Editorial Correction</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>- S21.MDL-1384 Issue 03,</td>
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<tr>
<td>10063291</td>
<td>STG AEROSPACE GERMANY GmbH</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>757-200-300, 737-800-900ER</td>
<td>REPLACEMENT SIDEWALL LIGHT LENSES AND END RETAINERS. Revision 1: Transfer of STC holder from STG Aerospace Limited to STG Aerospace GmbH</td>
<td>The Certification Basis for the original product as amended/adding additional or alternative airworthiness requirements: the following paragraphs(s) a) at a later amendment CS25 at Amendment 18CS 25.1, CS 25.25, CS 25.27, CS 25.301 (a)(i), CS 25.305 (a)(i), (c), CS 25.307 (a)(i), CS 25.561 (a)(i).</td>
<td>MASTER DATA LIST, Doc No: SG-1020-MDL-01, Rev No: IR. Instructions for Continued Airworthiness, Doc No: SG-1006-ICA-01, Rev No: IR for Boeing 737 INSTRUCTIONS FOR Continued Airworthiness, Doc No: SG-1006-ICA-01, Rev No: IR for Boeing 737 SER.</td>
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<td>10063292</td>
<td>DAC INTERNATIONAL INC.</td>
<td>FAA STC SA02365C</td>
<td></td>
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<td>Installation of a GDC 31 Roll Steering Converter</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>1. Master Drawing List No. 1049-200-XX, Rev. C, dated 23 August 2006, or later FAA approved revision. 2. Airplane Flight Manual 1049-2100-02, Rev. A, dated 7 October 2005, or later FAA approved revision.</td>
</tr>
<tr>
<td>10063304</td>
<td>AVIATION SPECIALTIES UNLIMITED, INC.</td>
<td>FAA</td>
<td></td>
<td></td>
<td></td>
<td>NVIS Lighting System installed as per MDL-900004-3 Rev L, dated 13 June 2017.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
<td>RollerCoast Flight Manual Supplement FMS-900004-6, rev C, dated 13 June 2016(+r), or later revisions of the above listed documents approved by EASA in accordance with EASA EC Decision 2004/190/EC (or subsequent revisions off this decision) and the Technical Implementation Procedures of EUVIAS Bilater</td>
</tr>
<tr>
<td>10063316</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td>A319-131, A319-132, A319-133, A319-114, A319-115, A319-111, A319-112, A319-113</td>
<td></td>
<td>-201 Antenna Doubler Installation. This revision will include -201 WADL antenna doubler installation, which was not considered by STC Twenty One Limited during the initial approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unamended and remain applicable to this certificate/approval.</td>
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<tr>
<td>STC Holder</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>ALPINE AEROTECH LP</td>
<td>Cargo Mirror &amp; Pulse Light Kit as per Master Data List AAL-290-014-001,Rev B</td>
<td>The Certification Basis for the original product as amended by EASA in accordance with the Technical Implementation Procedures of EASA Canadair S.1 Agreement; Installation instructions as applicable to each model...</td>
<td>Valid for all the types and models: RFMS AAL-290-015-001, initial issue, dated 21 Feb 2014</td>
<td>Prior to installation of this design change it must be determined that the...</td>
<td>Prior to installation of this design change it must be determined that...</td>
<td>09/10/2017</td>
<td>Active</td>
<td>10063340</td>
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<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>Cabin Interior Reconfiguration on an Airbus 320 aircraft.</td>
<td>The Certification Basis for the original product as amended by EASA in accordance with the Technical Implementation Procedures of EASA Canadair S.1 Agreement; Installation instructions as applicable to each model...</td>
<td>Valid for all the types and models: RFMS AAL-290-015-001, initial issue, dated 21 Feb 2014</td>
<td>Prior to installation of this design change it must be determined that the...</td>
<td>Prior to installation of this design change it must be determined that...</td>
<td>18/10/2017</td>
<td>Active</td>
<td>10063346</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>AS 350 D</td>
<td>The Certification Basis for the original product as amended by EASA in accordance with the Technical Implementation Procedures of EASA Canadair S.1 Agreement; Installation instructions as applicable to each model...</td>
<td>Valid for all the types and models: RFMS AAL-290-015-001, initial issue, dated 21 Feb 2014</td>
<td>Prior to installation of this design change it must be determined that the...</td>
<td>Prior to installation of this design change it must be determined that...</td>
<td>10/10/2017</td>
<td>Active</td>
<td>10063358</td>
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<tr>
<td>MAGNETIC MRO AS</td>
<td>Installation of Stretcher Assembly PN 1120030-310 into the passenger cabin of three aircraft types: B737-800 (B interior layout), B747-400 (7 interior layouts)</td>
<td>The Certification Basis for the original product as amended by EASA in accordance with the Technical Implementation Procedures of EASA Canadair S.1 Agreement; Installation instructions as applicable to each model...</td>
<td>Valid for all the types and models: RFMS AAL-290-015-001, initial issue, dated 21 Feb 2014</td>
<td>Prior to installation of this design change it must be determined that the...</td>
<td>Prior to installation of this design change it must be determined that...</td>
<td>19/09/2019</td>
<td>Active</td>
<td>10063360</td>
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<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>CAMERA SYSTEM INSTALLATION</td>
<td>The Certification Basis for the original product as amended by EASA in accordance with the Technical Implementation Procedures of EASA Canadair S.1 Agreement; Installation instructions as applicable to each model...</td>
<td>Valid for all the types and models: RFMS AAL-290-015-001, initial issue, dated 21 Feb 2014</td>
<td>Prior to installation of this design change it must be determined that the...</td>
<td>Prior to installation of this design change it must be determined that...</td>
<td>12/03/2020</td>
<td>Active</td>
<td>10063363</td>
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<td>STC No</td>
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<td>Rev.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>10063370</td>
<td>AEROCORP AVIONIC SOLUTIONS Inc.</td>
<td></td>
<td>DHC-6 SERIES 300</td>
<td>Installation of Emergency Cabin Lighting System.</td>
<td>This STC is approved only for the original product remains applicable to this certificate/approval.The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/10/2017</td>
</tr>
<tr>
<td>10063373</td>
<td>S4A SOLUTIONS FOR AVIATION, S.A.</td>
<td>60</td>
<td>Antenna installation provisions in Learjet 80</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>According to S4A Master Drawing List L6010-RPT02 Ed/or later revisions of the above listed documents approved by EASA To be installed i.e. w. S4A BB L6910-SB01 Ed/To be maintained a.e. S4A MM/S L6010-HDK2 Ed/1 or later Revisions</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/10/2017</td>
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<tr>
<td>10063383</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 1</td>
<td>A321-111/-212</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AFTM references AFMS- A320-17-015-001 &amp; AFMS- A320-17-015-002r or later revisions of the above listed documents approved by EASA Master Data List: Change Approval Sheet ref: CAS-A320-17-15 Rev 0.ICA: ICA- A320-17-015-001 Rev. 1.</td>
<td></td>
<td>1) This STC does not constitute installation approval of any aerial equipment. It only provides data that may partially support the implementations or activations. This STC may only be applied to an aircraft when an airborne base station using the cellular/or wireless technologies is concur...</td>
<td>16/10/2017</td>
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<tr>
<td>10063390</td>
<td>AVIATION TRADERS LIMITEDEUROPE A.S.</td>
<td>550</td>
<td>Installation of TCAS II with MST 67A Mode S TransponderRev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Definition Document DD/ 01/15/924 Issue SATL Flight Manual Supplement 24 Issue 1 &amp; AFM: references AFMS- A320-17-015-002r or later revisions of the above listed documents approved by EASA Master Data List: Change Approval Sheet ref: CAS-A320-17-15 Rev 0.ICA: ICA- A320-17-015-001 Rev. 1.</td>
<td></td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aerial/aircraft configuration shall bedetermined by installer. This certificate shall remain valid unless otherwise surrendered or revoked...</td>
<td>10/11/2011</td>
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<tr>
<td>10063394</td>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.</td>
<td></td>
<td>412, 412CF, 412EP</td>
<td>Installation of Enhanced Mode S TransponderRev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification Summary Sheet ATL/GEN/M027 Issue 3, dated November 5 2004r or later revisions of the above listed document(s) approved/acceptable under the EASA system.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aerial/aircraft configuration shall bedetermined by installer. This certificate shall remain valid unless otherwise surrendered or revoked...</td>
<td>10/11/2011</td>
</tr>
<tr>
<td>10063395</td>
<td>AVIATION TRADERS LIMITEDEUROPE A.S.</td>
<td>REV. 1</td>
<td>737-300</td>
<td>Installation of Enhanced Mode S TransponderRev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aviation Traders Ltd Definition Document DD/01/17/0168 Issue 1, dated October 27, 2004r or later revisions of the above listed document(s) approved/acceptable under the EASA system.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aerial/aircraft configuration shall bedetermined by installer. This certificate shall remain valid unless otherwise surrendered or revoked...</td>
<td>10/11/2011</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10063397</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td>ATP</td>
<td>Elementary and Enhanced Mode S Transponder Rev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aviation Traders Ltd Definition Document DD/0868/2007 Issue Aviation Traders Ltd AFM Supplement Number 21 Issue 1or later revisions of the above listed document (s) approved/accepted under the EASA system.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configuration shall be determined by installer. Prior to installation of this change/repair it must be determined that it is in...</td>
<td>10/11/2021</td>
</tr>
<tr>
<td>10063398</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td>737-800</td>
<td>Installation of Enhanced Mode S Transponder Rev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aviation Traders Ltd Definition Document DD/0159/2019 Issue Aviation Traders Ltd AFM Supplement Number 27 Issue 1 or later revisions of the above listed document (s) approved/accepted under the EASA system.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configuration shall be determined by installer. Prior to installation of this change/repair it must be determined that it is in...</td>
<td>10/11/2021</td>
</tr>
<tr>
<td>10063399</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 1</td>
<td>DC-8-63, DC-8-61, DC-8-63, DC-8-64</td>
<td>Introduction of a Honeywell EGPM5 MA VHF Rev. 1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aviation Traders Ltd Definition Document DD/0563/2007 Issue Aviation Traders Ltd AFM Supplement Number 22 Issue 1 or later revisions of the above listed document (s) approved/accepted under the EASA system.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configuration shall be determined by installer. Prior to installation of this change/repair it must be determined that it is in...</td>
<td>10/11/2021</td>
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<td>10063400</td>
<td>ROEDER PRECISION GMBH</td>
<td></td>
<td>114, 114A, 114B, 114TC, 112, 112TC</td>
<td>Installation of Bendix King Mode S Transponder KT 73</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>EASA STC 2004-11566 System and MST 67A Supp. CAS 67 TCAS 04th June 2004.- AFM</td>
<td>As listed in the AFMPrior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>16/10/2017</td>
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<tr>
<td>10063401</td>
<td>ROEDER PRECISION GMBH</td>
<td></td>
<td>M20, M20A, M20B, M20C, M20D, M20E, M20F, M20G, M20H, M20J, M20K, M20L, M20M, M20N, M20O, M20P, M20Q, M20R, M20S, M20T, M20U, M20V, M20W, M20X, M20Y</td>
<td>Installation of Bendix King Mode S Transponder KT 73</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. Approval as per EASA STC No.2004-12063 The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable.</td>
<td>EASA STC 2004-12063 System and MST 67A Supp. CAS 67 TCAS 04th June 2004.- AFM</td>
<td>As listed in the AFMPrior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/11/2017</td>
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<td>10063402</td>
<td>ROEDER PRECISION GMBH</td>
<td></td>
<td>SA227-TT, SA226-AT, SA227-AT, SA227-AC, SA227-BC, SA226-BC, SA226-7B, SA226-T, SA226-TC</td>
<td>Installation of TCAS II System and Mode S Transponder as per LBA STC No. SA1390 Issue 1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. Approval as per EASA STC No. 2004-5755 The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable.</td>
<td>EASA STC 2004-5755 System and MST 67A Mode S System for Fairchild Doc No. SA226/227/NY-1002-08, Rev 0, dated 21st May 2004 or later revisions of the above listed documents approved by...</td>
<td>As listed in the AFMPrior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/11/2017</td>
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10063401 
ROEDER PRAEZISION GMBH 

114TC, 114A, 114B, 112, 112TC 

Installation of Beneteau King Mode S Transponder KT 73 

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. Approval as per EASA STC No. 2004-5765. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable.

As listed in the ATM. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10/12/2017 
Active 
10000402

10063405 
ATELIER CHAUBORD REV. 1 

DGAC SF-0001 


Installation of an exhaust with a silencer.

The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

12/11/2017 
Active 
10000405

10063406 
FARGO MANUFACTURING CO. REV. 1 

FAA SF6637GL 


Installation of Auxiliary Fuel System

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval without any...

Prior to installation of this design change/repair it must be determined that the interrelationship between this design change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

19/06/2020 
Active 
10000406

10063409 
ROEDER PRAEZISION GMBH 

M20/M20A/M20B/M20C/ M20D, M20G/M20M/M20S M20/M20F/M20G/M20J/ M20K 

Installation of GNS 430 WAAS navigation system.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

As stated in limitation section NAYAK AFM & Document No. GNS430-NY-05131- NY-05131-05. AFM 0 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

15/11/2017 
Active 
10000409

10063410 
AIRBUS HELICOPTERS UK LIMITED 

CAA UK RESPONSIBILITY 

MBB-BK117 D-2 

Hoist Controller and Reel In/Out Switch.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

Only applicable if OEM hoist system, AH Change Nr. E-2024, is installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

12/10/2017 
Active 
10000410

10063415 
MT-PROPELLER ENTWICKLUNG GmbH 

P2010 


According to MT-Propeller Master Documentation List No. E-2982, Rev 1:(<>) To be operated in accordance with MT-Propeller AFM & Document No. E-2979, Rev 2:(<>) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

12/10/2017 
Active 
10000415
<table>
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<tr>
<th>STC Holder</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>NORTWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>Cabin Interior Reconfiguration - AUA 767</td>
<td>Certification Summary Doc Nr N0594-D00-1 Rev - Master Drawing List Doc Nr N0594-D00-1 Rev Conf, PTC (HR) and SD for new seat install H4-01 Enhanced AIRworthiness Programme for Aerospace systems. The following paragraph(s) at a later amendment:..</td>
<td>Certification Summary Doc Nr N0594-D00-1 Rev - Master Drawing List Doc Nr N0594-D00-1 Rev Conf, PTC (HR) and SD for new seat install H4-01 Enhanced AIRworthiness Programme for Aerospace systems. The following paragraph(s) at a later amendment:..</td>
<td>Limited to Boeing 767-300, S/N 25273, 26417, 27095, 27098, 27099, 28328, 30331 - 30383LH..</td>
<td>16/10/2017</td>
<td>Active</td>
<td>10063425</td>
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<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>Installation of Rockwell Collins Pro Line DFS MOD 7082A : Falcon 7X with M1000</td>
<td>Installation of Rockwell Collins Pro Line DFS MOD 7082A : Falcon 7X with M1000</td>
<td>1. Master Data List S75-00-001, revision IR, dated 28 August 2017. 2. Airplane Flight Manual Supplement S75-00-0055, revision (RV)() dated 23 August 2017. 3. Airplane Flight Manual Supplement S75-00-0088, Revision IR, dated 04 March 2018 for optional FMSv4.2. 4. Instructions for Continued AIRworthiness...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the AIRworthiness of the product..</td>
<td>30/09/2022</td>
<td>Active</td>
<td>10063435</td>
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<td>DASSAULT FALCON SERVICE</td>
<td>Installation of a Honeywell ASPIRE 200 - SATCOM system.</td>
<td>The Certification Basis for the original product remains applicable to all certificates/ approvals, except where amended by additional or later amendments as indicated on FAA STC (SA1221S) The requirements for environmental protection and the associated controlled noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval..</td>
<td>Instructions for Continued AIRworthiness DFS-DE-15-7X-7082-01 Issue dated 09 October 2017(&gt;&gt;&lt;&lt;) or later revisions of the above listed documents approved by EASA Master Document List DFS-DE-17-7X-7082 Issue A dated 13 October 2017.</td>
<td>This STC is applicable to aircraft Falcon 7X post-modification M-0000/0 N405. This design change must be applied concurrently with Dassault Aviation modification MOD-DE-15-7X-7082-01. Prior to installation of this design change it must be determined that the interrelationship between this design change..</td>
<td>13/10/2017</td>
<td>Active</td>
<td>10063451</td>
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<tr>
<td>HONEYWELL INTERNATIONAL, Inc.</td>
<td>Installation of a Honeywell ASPIRE 200 SATCOM system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated controlled noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List (MDL) MOD-AS350ASPIRE200-5201, Revision B, dated 13 July 2017, or later Federal Aviation Administration (FAA) approved revision. Rotorcraft must be maintained in accordance with Instructions for Continued AIRworthiness (ICA) MOD-AS350ASPIRE200-5201, Revision A, dated 23 May 2017 or later..</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the AIRworthiness of the product..</td>
<td>16/10/2017</td>
<td>Active</td>
<td>10063452</td>
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<td>SECURAPLANE TECHNOLOGIES, Inc.</td>
<td>Installation of Securaplane Sealed Lead Acid Batteries.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s) (D-GEN02) PTC (HR) and SD for new seat install H4-01 Enhanced AIRworthiness Programme for Aerospace systems. The following paragraph(s) at a later amendment:..</td>
<td>Master Data List AM0301-DL02-00, Revision, dated 19 July 2017, or later Federal Aviation Administration (FAA) approved revision. The modified aircraft must be maintained in accordance with the instructions for Continued AIRworthiness AM0301-DL02-00, Revision, dated 19 May 2017 or later FAA accepted..</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the AIRworthiness of the product..</td>
<td>19/10/2017</td>
<td>Active</td>
<td>10063459</td>
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<td>EASA STC no.</td>
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<td>STC Ref.</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10063461</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>500X 550 S550</td>
<td></td>
<td>GTN upgrade for LPV and TAWS A Operation.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements.</td>
<td>Installation in accordance with SCANDINAVIAN AVIONICS design modification 2108486. Master Document list 2108486-DOC.is 1 dated 13.10.2017. Operation in accordance with SCANDINAVIAN AVIONICS design AFMSupplement: 2108486-AM A, is 1 dated 28.08.2017. TAWS A: 2108486-AM B, is 2 dated 04.10.2017.</td>
<td>Prior to installation of this design change it must be determined that the installation in accordance with this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10063463</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>AS 350 B2/ A3 350 B3</td>
<td>FAA STC SRT095MRC</td>
<td>Installation of AH multi-bloc Center Console.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate. Approval of the requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unannounced and remain applicable to this certificate/ approval.</td>
<td>Airbus Helicopters, Inc. Master Drawing List (MDL) 350A02-1752. Rev A dated 28.08.2017. FAA approved revision. Instructions for Continued Airworthiness (ICA) 350MMS-035. Rev IR dated 21 September 2015 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the installation in accordance with this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10063477</td>
<td>ANCRA INTERNATIONAL LLC</td>
<td></td>
<td>FAA STC ST0898LA</td>
<td>Installation of main deck cargo loading system.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Regulations at Am/(25-0): 25.601, 25.609 (r-2), 25.621, 25.1541, Am11(25-23), 25.30(a)(4), 25.30, 25.471, 25.607, 25.619. Am11(25-46), 25.603, 25.605, Am11(25-51).</td>
<td>Ancra International LLC Controlled Data List, DLB6450, Revision F. dated 04 May 2017. Instructions for Continued Airworthiness (ICA). Ancra International LLC Document No. 10472, Revision B. dated 24 May 2017, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the installation in accordance with this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10063478</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td></td>
<td>FAA STC ST0297SE</td>
<td>First Class Cabin Refresh of Cathay Pacific Aircraft.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: for the following paragraph(s) at a later amendment: 25.553(a)(a)(a) at CS-25 Amendment 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unannounced and remain applicable to this certificate/approval.</td>
<td>Master Data List Aerospace Technologies, Inc. document No. N0551- D010 Revision B, dated 24 July 2017. Instructions for Continued Airworthiness, Northwest Aerospace Technologies, Inc. document No. N0551-070, Revision B, dated 2 August 2017 or later revisions of the above listed documents/approval.</td>
<td>The modification approval is limited to aircraft serial numbers determined by Northwest Aerospace Technologies, Inc. drawing 551N100. If the holder agrees to permit another person to use this certificateholder to perform the work, the holder shall give the other person written evidence of that permission.</td>
</tr>
<tr>
<td>10063487</td>
<td>E.A.D AEROSPACe AIRworthness</td>
<td></td>
<td>Rev.</td>
<td>A340-211, A340-212, A340-213</td>
<td>Satcom KA band &amp; Cabin WiFi Installation.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements Equivalent Safety Findings(s): CRI B-0301 (Vibration/buffeting compliancetest for large external antenna installation). The requirements for environmental protection and...</td>
<td>- Master Drawing List, EAD Aerospace document No. 01501-EAS-A. MDL.Revision B, dated 17 October 2017.</td>
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<td>Rev. level</td>
<td>Foreign/NAA Code</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>BABCOCK MISSIONITICAL SERVICES DESIG N AND COMPLETIONS LIMITED</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>EC135 T2+</td>
<td>BABCOCK/MOD/134 EC135 T2+ Description: This STC takes an aircraft equipped with the Airbus factoryNVIS options and role equipped for air ambulance work, and re-structures it for air ambulance service. NVIS certification of the rework is carried out as part of this Change in order to maintain...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 27.Amdt 3 Appendix ACS 27 Amdt 3 Appendix BCS 27 Amdt 3 Appendix CCS-ACNS.B.VCS.020/ ACNS.B.VCS.025/ ACNS.B.VCS.030 The requirements for environmental protection...</td>
<td>Flight Manual Supplement FMS/134-002, Issue 2, dated 22 September 2017 or later revisions of the above listed documents approved by EASA—Design Report DR134-001, Issue 3, dated 22 February 2017— Compliance Checklist CCL/134-003 Issue 2, dated 20 July 2017— Compliance Checklist CCL/134-004 Issue 2—.</td>
<td>1) This is a series Change applicable to all EC135 T2+ aircraft equipped with the modifications listed below and subject to compatibility with existing Changes: OEM Changes—Airbus Helicopters Emergency Firefighting System. Ref Code L321M801057Third Party Changes—NVIS Curtains. Ref Mod No.: MCH/A135...</td>
<td>20/10/2017</td>
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<tr>
<td>LUFTHANSA TECHIN AG</td>
<td>REV. 3</td>
<td>A321-253NX/-271N A320-232 A321-231/-232 A320-232/-251N A320-232 A320-232 A320-232 A319-131-1-130 A319-131-1-132 A319-131-1-132</td>
<td>Provisions for large Ka-Band antenna and systems installation (Rev. 2 extends the applicability to the A321-253N/Rev. 3 extends the applicability to A319-133; A320-214/-217/11A321-211/-212/212/271N)</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Finding(s): B-101 “Vibration/ buffeting compliance criteria for large external antenna components” The requirements for environmental protection...</td>
<td>Installation allowed only on aircraft that satisfy the criteria and configuration defined in Description of Change AVA01-23 Rev 26 and STRConfiguration Sheet AVA02-63-004 Rev.6 or later revisions of these documents as applicable/Prior installation of this change/repair it must be determined that...</td>
<td>03/08/2022</td>
<td>Active</td>
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<tr>
<td>ECMS AVIATION SYSTEMS GMBH</td>
<td>E0635 T3 E0635 P3/E0635 T1/ E0635 T2+ E0635 T2+ EC135 T2+ EC135 T2+ EC135 P3 1/ E0635 P2+</td>
<td>External Load Attachment Device EA0203 Left hand, right hand or both sided installation of External Loadattachment Device on existing hard points to attach up to four loads.</td>
<td>The External Load Attachment Device for EADS3. Left hand, right hand or both sided installation of External Loadattachment Device on existing hard points to attach up to four loads.</td>
<td>Flight Manual Supplement for External Load System. EA003 or EC135 P1/P2; P2+; P3; T1; T2; T2+; T3; E0635 P2+; P3; T1; T2; T3; Original Issue (RO) Airworthiness Limitation Section Chapter 11 of the Component Maintenance Manual and Instructions for Continued Airworthiness (CMICMA) for External Load Sys...</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>20/10/2017</td>
<td>Active</td>
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<tr>
<td>SOCIETE DUC A/DUC HELICES PROPPELLERS</td>
<td>REV. 2</td>
<td>TB 9; TB 20; TB 10; TB 20; TB 21</td>
<td>TBs Spinner/Backplane SIC update (Backplane screw change)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>On TB STOMC 31 A dated 26/06/2019 or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>27/06/2019</td>
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<td>STC Holder</td>
<td>Integrated Global Navigation System</td>
<td>10063567</td>
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<td>Limitations/Conditions</td>
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<td></td>
<td>Installation of Rockwell Collins TP9-962</td>
<td>Automatic Dependent Surveillance - Broadcast (ADS-B) Out Transponder(s).</td>
<td></td>
<td>1. Master Data List (MDL), 216307-MLD, Rev. B, dated 17 August 2017, ord. FAA approved revision.2. Instructions for Continued Airworthiness (ICA) Document 216307-ICA, Rev. A, dated 19 May 2017(&lt;&gt;) or later FAA accepted revision.</td>
<td>Prior to installation of this design change must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce adverse effect upon the airworthiness of the product.</td>
<td>23/10/2017</td>
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<th>STC Holder</th>
<th>Innovative Solutionsand Support, Inc.</th>
<th>10063557</th>
<th>Active</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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| STC Holder | Lufthansa Technik AG | 10063559 | REV. 2 | Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any imp... | - Supplemental Type Definition Document, Ref. STDCG-3170G, Issue 08Rev 00- Aircraft Flight Manual Supplement, Ref. A-00-15/93-AFM Issue 01 Rev 01- Instructions for Continued Airworthiness, Ref. A-00-15/93-AAM Issue 01Rev 00- Flight Crew Data, Ref. A-00-15/93-FCD Issue 01 Rev 02later revisions o. | - Only applicable to aircraft with EASA STC 10063837 installed - Installation on aircraft models only that satisfy the configurations criteria defined in Description of Change AVF1-25 Rev 34 or laterapproved revisions of this documentation prior to installation of this design change it must be determin... | 19/08/2022 | Active | 10063559 |

| STC Holder | Lindstrand Technologies Ltd. | 10063567 | CAA UK RESPONSIBILITIY | Certification Basis for the original product as amended following additional or alternative airworthiness requirements:thesefollowing paragraph(s) at a later amendment: CS-31HB Amdt. 1The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any imp... | None | | None | 24/10/2017 | Active | 10063567 |

| STC Holder | Sabena Technics BOC | 10063572 | A330-203 | Certification Basis for the original product as amended following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: Applicable regulations at CS 25 Amendment 4 for details reinitialized Associated Certification Plan ref 4802-3A-CP-R02, section 4.5 | AFM ref 4802-02-AAFMS rev 10 late r revisions of the above listed documents approved by EASA, Master Data List ref 4802-3-AAML-000-ICA ref 4802-03-AICA-000 | Refer to the concurrent requirements of the Certification Plan ref 4802-03-A-CP-R02, section 2.5. Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce adverse effect upon the airworthiness of the product. | 25/10/2017 | Active | 10063572 |

| STC Holder | Aerospace Logic Inc. | 10063579 | STC Holder | Certification Basis for the original product as amended following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: Applicable regulations at CS 25 Amendment 4 for details reinitialized Associated Certification Plan ref 4802-3A-CP-R02, section 4.5 | - Installation as primary or secondary, new or replacement must be in accordance with Aerospace Logic Certified Installation List - Document S200-CLL, Rev. 1.0 dated 02/12/2009 (Operation of transport Canada on 14/12/2009). | Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce adverse effect upon the airworthiness of the product. | 25/10/2017 | Active | 10063579 |
The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Installation in accordance with RAISBECK Engineering MasterDrawing List(MDL). 14BC-61-000, Revision A, dated August 28, 2017. This installation requires the prerequisite installation of STC SA5151NM(LBA STC SA1011) which installs Raisbeck Dual Air Body Strakes. Aircraft must be operated in accordance with the above listed documents or later revisions of the above listed documents approved by EASA.

The Certification Basis for the original product remains applicable to this certificate/ approval. This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Installation in cockpit, on both Captain and F/O sides, of one 28VDC converter dual USB charging plugs.

The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

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<th>Date of Issue</th>
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<tr>
<td>10063637</td>
<td>AVOX SYSTEMS Inc</td>
<td>FAA STC</td>
<td>ST0180W</td>
<td>A321-213/-231/-232</td>
<td>A321-111/-112/-131/-111/-212</td>
<td>Installation of the AVOX LavOx Passenger Lavatory Oxygen System.</td>
<td>None.</td>
<td>06/11/2017</td>
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<td>10063638</td>
<td>LUTHERANS TECHNIX AG</td>
<td>FAA STC</td>
<td>ST0180W</td>
<td>A321-213/-231/-232</td>
<td>A321-111/-112/-131/-111/-212</td>
<td>ADS-B Out Activation</td>
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<td>10063641</td>
<td>AVIONITEC AG</td>
<td>FAA STC</td>
<td>ST0180W</td>
<td>777-200</td>
<td>340A</td>
<td>Avionics renewal Removal- Bendix/King KI 520A Analog HSI with GARMIN A3000 with Weather Radar and MAGNAVAC Mobile System.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/11/2017</td>
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<tr>
<td>10063684</td>
<td>PHOENIX AEROSPACE LIMITED</td>
<td>FAA STC</td>
<td>ST0180W</td>
<td>S-61N</td>
<td>S-61N</td>
<td>Introduction of Sandel ST3400H HTAWS and SG102D AHR, Rev. 1-Addition of S-61N serial number 61809</td>
<td>Limited to serial numbers 61269, 61270 and 61809. Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/10/2019</td>
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<td>Date of Issue</td>
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**List of EASA Supplemental Type Certificates**

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<th>EASA STC no.</th>
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<th>Rev. level</th>
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<th>Limitations/Conditions</th>
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<tr>
<td>10083770</td>
<td>BELL</td>
<td>0</td>
<td>SR0913RC</td>
<td>505</td>
<td>Installation of an Automatic Door Opener Kit</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Data, EASA Report No. AA-16060 Revision A(4) - (in dated 11 April 2017, or later FAA approved revision 2. Instructions for Continued Airworthiness, Report No AA-15071(4) - (in dated 8 dated 30 March 2017, or later FAA accepted revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/12/2017</td>
<td>Active</td>
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<td>10083774</td>
<td>SAAB AB</td>
<td>REV. 2</td>
<td>B200GT</td>
<td></td>
<td>Installation of SATCOM using Iridium Transfer to SAAB ABKTEBOLG. Rev. 2: Correction of DOA reference.</td>
<td>The Type and OS2 Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Installation Data TDM17-24-3-D Revision A - Instructions for Continued Airworthiness TDM17-24-4-ICA Revision A - Certification Program TDM17-24-1-CP Revision A - Design Assessment TDM17-24-1-DA Issue A - Ground Test TDM18-35-1-GT Revision A - Generic Interference Test TDM18-35-1-GT(1) - First Artic...</td>
<td>This change is only applicable to B200 GT with Fusion avionics. Applicable as: 1. No IF is installed Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/12/2023</td>
<td>Active</td>
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<td>10083786</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST03161CH-D</td>
<td>CL 650-1A11/-2A12/-2B16 BD-700-1A10/BD-700-1A11</td>
<td>Use of Portable Electronic Devices (PEDs) to Control installed Aircraft Systems.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- JET AVIATION ST. LOUIS INC. INDEX LIST No. 35-406003 REV CATED 01/December 2014.</td>
<td>The Wi-Fi and Cabin Management Systems on which the PEDs will function require EASA approval by means other than this STC.</td>
<td>27/11/2017</td>
<td>Active</td>
<td>10083786</td>
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<tr>
<td>10083787</td>
<td>AIRBUS HELICOPTERS DEUTSCHLAND GmbH</td>
<td></td>
<td>EC135 T2+; EC135 T3; EC135 P3; EC135 T1, EC135 T2; EC135 P1; EC135 P52; EC135 P2B</td>
<td></td>
<td>Installation of a Cockpit Voice Flight Data Recorder (CVFDR) System in accordance with Compliance Report SK-MAC-117-ORP-B-1385.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement CVFDR Upgrade SK-MAC-117 SUP-B-1375; Issue A, or later revisions of the above listed document are approved by EASA Compliance Report SK-MAC-117-ORP-B-1385, Issue D; Certification Manual SK-MAC-117-C6A-B-1375; Issue C...</td>
<td>Applicable to variants with CPCS installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/11/2017</td>
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<td>10063707</td>
<td>ELBE FLUGZEUGWERK E GmbH</td>
<td>REV. 3</td>
<td>A330-342, A330-343, A330-323, A330-321, A330-320, A330-301, A330-300, A330-201, A330-202, A330-203, A330-202, A330-201</td>
<td>Powered Cargo Loading System Installation in A330 Passenger Freight/This approval covers the installations of powered cargo loading systems A330-200 and A330-300 Passenger to Freighter models.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Top Engineering Order EFW-EO-0048-001 Revision D- Type Investigation EFW-TIP-0048-001 Revision E-Instructions for Continued Airworthiness EFW-E-ED-0527 Revision B Mode PSF Modification certified with EFW-EU-0198-001 Rev. A and EFWE-TIP-0598-001 Rev. B- EFWE-85-0098 R00- Modification of Cargo Lo...</td>
<td>This modification must be installed with the following related changes: EFWF - A330 Passenger to Freighter. Approval number 10063798.Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any otherwise installed change...</td>
<td>22/02/2023</td>
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<td>10063708</td>
<td>ELBE FLUGZEUGWERK E GmbH</td>
<td>REV. 33</td>
<td>A330-342, A330-343, A330-323, A330-321, A330-320, A330-301, A330-300, A330-201, A330-202, A330-203</td>
<td>Airbus A330 Passenger to Freighter conversion Rev. 16 for all airworthiness updates and the introduction of A3050PS Smoke Curtain Design.</td>
<td>The Type and OSD Certification Bases for the original product are amended by the following additional or alternative airworthiness requirements. Special Condition (s): As per STDS1 10063798 at revision 9 with following special conditions: D-02 Courier Compartment D-03 Fire Protection of Essential systems/...</td>
<td></td>
<td></td>
<td>This modification must be installed with the following associated changes: Power Cargo Loading System per EASA STC 10063797b. Couryer Module, including the following options, per EASA STC 10063798.</td>
<td>31/05/2024</td>
<td>Active</td>
<td>10063708</td>
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<tr>
<td>10063796</td>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>FAA SR112205C</td>
<td>EC 120 B</td>
<td>Installation of Helias S Stability Augmentation System and Autopilot(SAS/ AP).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. Special Condition (s): CRI CS 27.1309 (FAA Special Condition 27-040-SC). The requirements for environmental protection and the associated certificated noise and/or emissions levels remain unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>23/11/2017</td>
<td>Active</td>
<td>10063796</td>
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<td>10063802</td>
<td>CARSON HELICOPTERS, INC.</td>
<td>FAA STC SR0257NY</td>
<td>SEE FAA AML</td>
<td>Installation of Goodrich AC rescue hoist system P/N 44311-10-2NEW-PAGE</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td></td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/03/2018</td>
<td>Active</td>
<td>10063802</td>
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<tr>
<td>10063824</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACS5)</td>
<td>FAA STC ST02632LA</td>
<td>SEE FAA AML</td>
<td>Modification to replace the existing Mode S Transponder system with ACSS/XT-700 Mode S Transponder with ADS-B Out and optionally install theACS5 NGST-900 GP as a position source for ADS-B. Out Rev. 3 includes an added NKT-700 transponder part number additional certified aircraft models and addi...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA approved AML STC Master Data List (MDL) as defined, along with the associated FAA approved Airplane Flight Manual Supplement and Instructions for Continued Airworthiness in the AML No. ST02632LA, dated 09 June 2017, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The STC is valid for those aircraft types and mo...</td>
<td>03/09/2018</td>
<td>Active</td>
<td>10063824</td>
<td></td>
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</tbody>
</table>
List of EASA Supplemental Type Certificates

10063836
**JET AVIATION ST. LOUIS, Inc.**
**REV. 1**
**FAA ST02600CH-D**
**BD-700-1A10**

Installation of a Pocket Door.

The Certification Basis for the original product as amended/modified: additional or alternative airworthiness requirements defined in Document 0-01 Installation of a Pocket Door. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

**Certification basis:**

- **EASA STC No. TBD (EASA 1)**
- **FAA STC No. ST02620CH-D**
- **BD-700-1A10**
- **Canadian Type Certificate A-205**

**Limitations/Conditions:**

- 1. Restricted to ASN 9000.2 Door finish requirements and installation environment must adhere to applicable flammability and/or decompression requirements. Refer to Index List for requirements definition 3. Maximum installed door weight must not exceed the requirements defined by the companion product.

**Date of Issue:** 10/01/2018
**Status:** Active

10063837
**JET AVIATION ST. LOUIS, Inc.**
**FAA STC ST04157CH**
**BD-700-1A10**

Modification of an Executive Interior Arrangement.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

**Certification basis:**

- **EASA STC No. TBD (EASA 1)**
- **FAA STC No. ST04157CH**
- **BD-700-1A10**
- **Canadian Type Certificate A-205**

**Limitations/Conditions:**

- 1. Restricted to ASN 9000.2. EASA STC No. 1004418 Rev 1 is prerequisite for this STC.
- 3. EASA STC No. TBD (EASA PN 0010494940) is a prerequisite for this STC.

**Date of Issue:** 23/11/2017
**Status:** Active

10063838
**POLYTECH AERO SUPPORT INC.**
**TCCA STC SA17-37**

A330 CABIN LAYOUT RECONFIGURATION: This modification reconfigures the passenger cabin of Airbus A330 Aircrakft type, certified under Canadian Type Certificate A-205 applicable. The reconfigured cabin includes: new J/C seats (12) and Y/C seats (33) interior placards, emergency equipment (rearranged to...)

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

**Certification basis:**

- **EASA STC No. TBD (EASA 1)**
- **FAA STC No. ST02620CH-D**
- **BD-700-1A10**
- **Canadian Type Certificate A-205**

**Limitations/Conditions:**

- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a loss of performance and/or reliability of the product. The aircraft should be weighed upon modification...

**Date of Issue:** 23/11/2017
**Status:** Active

10063839
**MAP AIRCRAFT PART 21 A9**

AS 365 N3

A330 PN0 Installation.

The Certification Basis for the original product as amended/modified: the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment, CS-29 Amendment 3.20, 29.301, 29.305, 29.307, 29.309, 29.561, 29.601, 29.603, 29.605, 29.609, 29.610, 29.613, 29.629, 29...

Flight Manual Supplement MAP084FM Issue C dated 01/11/2017< or later revisions of the above listed documents approved by EASA. Reference (Oil) MAP084K010 Issue C dated 17/11/2017.

**Certification basis:**

- **EASA STC No. TBD (EASA 1)**
- **FAA STC No. ST02620CH-D**
- **BD-700-1A10**
- **Canadian Type Certificate A-205**

**Limitations/Conditions:**

- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a loss of performance and/or reliability of the product. The aircraft should be weighed upon modification...

**Date of Issue:** 23/11/2017
**Status:** Active

10063854
**HELIKOPTER AIR TRANSPORT GESSELLSCHAm.b. H.**
**REV. 5**

EC135 T3 EC135 T2+ EC135 T3 EC135 T2 EC135 T1 EC135 P3 EC135 P2 EC135 P2 EC135 P1 EC135 P2+ EC135 P1

Inst. of mutipurpose interior lining (Configuration 25-001). Rev. 1 included the Medical Attachment Structure configuration 25-002Rev. 2 required changes in the forward aircraft roof sections / attachment structure due to added adaptability to new introduced EC135 T3P3 Helionix helicoptersRev. 3 ens...


**Certification basis:**

- **EASA STC No. TBD (EASA 1)**
- **FAA STC No. ST02620CH-D**
- **BD-700-1A10**
- **Canadian Type Certificate A-205**

**Limitations/Conditions:**

- Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a loss of performance and/or reliability of the product.

**Date of Issue:** 07/04/2022
**Status:** Active

10063830
**JET AVIATION ST. LOUIS, Inc.**
**FAA STC ST04157CH**
**BD-700-1A10**

Installation of a Pocket Door.

The Certification Basis for the original product as amended/modified: additional or alternative airworthiness requirements defined in Document 0-01 Installation of a Pocket Door. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

**Certification basis:**

- **EASA STC No. TBD (EASA 1)**
- **FAA STC No. ST02620CH-D**
- **BD-700-1A10**
- **Canadian Type Certificate A-205**

**Limitations/Conditions:**

- 1. Restricted to ASN 9000.2 Door finish requirements and installation environment must adhere to applicable flammability and/or decompression requirements. Refer to Index List for requirements definition. 3. Maximum installed door weight must not exceed the requirements defined by the companion product.

**Date of Issue:** 10/01/2018
**Status:** Active
<table>
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<tr>
<th>STC Holder</th>
<th>System</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Issue</th>
<th>Record</th>
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</thead>
<tbody>
<tr>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>REV. 5</td>
<td></td>
<td></td>
<td>A321-271N-2717X</td>
<td>Activation of the Airborne Complementary Ground Component (ACGC) System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/revision.</td>
<td>Master Document List, EAD Aerospace document No. 01538-EAS-A MDL, Revision F, dated 29 March 2019, Airplane Flight Manual Supplement, EAD Aerospace document No. 01538-EAS-A/AFM-01, Revision B, dated 17 July 2018, for Continued Airworthiness, EAD Aerospace document No. 01538-EAS-A/ICA-01, Rev...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>2019</td>
<td>Active</td>
<td>10025307</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
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<td></td>
<td>A330-343</td>
<td>A330 DLH 2 Class Configuration - PAC</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/revision.</td>
<td>Supplemental Type Definition Document, Ref. STDD00-1627M, Issue 1, Rev. 0, Compliance Record List, Ref. CRL00-1627M, Issue 1, Rev. 0, Airplane Flight Manual Supplement, Ref. A-00-17-20/M, Issue 1, Rev. 0, Instructions for Continued Airworthiness, Ref. A-00-17-20/IC, Issue 3, Rev. 0, or later revisions...</td>
<td>The following pre-requisite installations apply: - Installation of 1000-series photo luminescent emergency path markingsystem certified by EASA STC 1004240 &amp; Minor Changes Approval A/AFM-555/ - Installation of ZNAT Interior Reconfiguration system certified by EASA STC 10059278 &amp; Minor Changes Approval N54/...</td>
<td>2017</td>
<td>Active</td>
<td>10025308</td>
<td></td>
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<tr>
<td>LITEAR AVIATION PRODUCTS Inc.</td>
<td>FAA STC ST002388E</td>
<td></td>
<td></td>
<td>ATR 72-212 A</td>
<td>Installation of lightweight window plugs.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/revision.</td>
<td>Litear Document LAP2012, Revision 9, dated 11 January 2017 or later, FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>2017</td>
<td>Active</td>
<td>10025310</td>
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<tr>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC ST002388E</td>
<td></td>
<td></td>
<td>A330-203/ A330-303</td>
<td>Installation of monuments and structural provisions on Airbus Model A330-200/300 series airplanes.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/revision.</td>
<td>B/E Aerospace, Inc. - FSI (B/E - FSI) Master Drawing List (MDL) BIFS6538-D01, Revision E, dated 24 November 2017 or later FAA approved revision. The Supplemental Instructions for Continued Airworthiness, B/E, FSDocument BIFS6538-D15, Revision R, dated 21 November 2017, or later FAA approved revision.</td>
<td>This STC certifies structural provisions for the installation of an interplane reconfiguration as defined in B/E, FSI Document BIFS6538-D41, Rev. C, Dated 24 October 2017 or later FAA approved revision. The FAA must evaluate and provide additional approval for the installation of the interior to ensure it...</td>
<td>2017</td>
<td>Active</td>
<td>10025311</td>
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<tr>
<td>SEE EASA-APPROVED MODEL LIST</td>
<td></td>
<td></td>
<td></td>
<td></td>
<td>Installation of ADR compliant equipment for Worthington cylinders in accordance with Schroeder Fire Balloons Document List TM00, Issue 1, dated 30 November 2017</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/revision.</td>
<td>- Schroder Fire Balloons Document List TM00, Issue 1, dated 30 November 2017, for Continued Airworthiness, EAD Aircraft document No. 01538-EAS-A STDD00-1627M, Issue 1, Rev. 0, Compliance Record List, Ref. CRL00-1627M, Issue 1, Rev. 0, Airplane Flight Manual Supplement, Ref. A-00-17-20/M, Issue 1, Rev. 0, Instructions for Continued Airworthiness, Ref. A-00-17-20/IC, Issue 3, Rev. 0, or later revisions...</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>2017</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>Gulfstream Aerospace Corporation</td>
<td>Installation of a Custom Interior</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/revision.</td>
<td>Gulfstream Aerospace Corporation Index List GC420008030, Revision C, dated 10 November 2005, or later FAA approved revision.</td>
<td>Descriptive data pertaining to this design change are consistent with that for duplication in other products. This approval is limited to only the installation made in Gulfstream GIV-X, Serial Number 4024. This STC does not permit manufacturing of parts for multiplane installations. The following STCs and...</td>
<td>04/12/2017</td>
<td>Active</td>
<td>10063941</td>
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<td></td>
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<tr>
<td>Sabena Technics BOD</td>
<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/revision.</td>
<td>- Master Data List, ref: MDL-02-0414, Issue A, dated 24 June 2021; - Aircraft Flight Manual Supplement, ref: AFMS- F3578-00-031, revision 1, dated 27 October 2017; - Instructions for Continued Airworthiness, ref: ICA- F3578-00-009, issued 17 November 2017; - Airworthiness Limitation Supplement,...</td>
<td>The following modifications must be concurrently embodied with the modification: - STC 10014507 Revision 2 – Floor Pathmarking installation (or later EASA approved revision).</td>
<td>01/12/2017</td>
<td>Active</td>
<td>10063950</td>
<td></td>
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| HaeCo Cabin Solutions, LLC | Installation of Collins SAT-906 SATCOM System. | The Certification Basis (CB) for the original product remains applicable to this certificate/revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/revision. | TIMCO Master Data List Number MDL-01189, Revision D, dated 25 February 1994, or later FAA approved revisions - Page Avjet Report No. PR-94-108, dated 16 March 1994(=)
Continual Maintenance Inspection Requirements... | Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 08/12/2017 | Active | 10063956 |
Revision D(=)
dated 2 June 2010 or later FAA approved drawing revisions - Instructions for Continued Airworthiness (ICA), EMTEQ Engineering Document No. 2-46336-905, Rev. B, dated 21 June 2010(=)
or later FAA approved revision. | Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 05/12/2017 | Active | 10063969 |
<p>| Butler National Corporation | The installation of dual Garmin GTX-35ES Mode-S transponders and single dual Garmin GTN 650/750 panel mounted navigation and global positioning system (GPS) units, or dual Garmin GTX-335R and/or GTX-35ER Mode-S transponders and single dual Garmin GTN 650/750/825/725 panel mounted navigation and GPS. | The Certification Basis (CB) for the original product remains applicable to this certificate/revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/revision. | Installed in accordance with Butler National Corporation's master detailed (MD) as specified on AML ST02235SE... and maintained in accordance with instructions for Continuing... | Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The STC is valid for those aircraft types and mo... | 05/12/2017 | Active | 10063971 |</p>
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<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC (n)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>1006374</td>
<td>BUTLER NATIONAL CORPORATION</td>
<td>FAA STC 204655E</td>
<td>SEE FAA AML</td>
<td>Installation of Dual Garmin GTX-3000 Transponders. GDL-88 Automatic/Dependent Surveillance-Broadcast (ADS-B)(h) with Wide Area Augmentation System/WAAS Global Positioning System (GPS) sensor and Flight Stream210 Bluetooth Connectivity installed for ADS-S/(Out) Compliance.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Butler National Corporation's Master Data List (MDL) as specified on AMLST3995555M. Maintained in accordance with Butler National Corporation/Instructions for Continued Airworthiness (ICA) as specified on AMLST3995555M. - Operating. In accordance with Butler National Corporation/Airline Flight Manuals.</td>
<td>Prior to installation of this design change it must be determined that the integration between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the other products. The STC is valid for those aircraft types and mo...</td>
<td>02/12/2017</td>
<td>Active</td>
<td>10020379</td>
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<tr>
<td>1006378</td>
<td>AIRWORK &amp; HELISELEREI GmbH</td>
<td>REV. 2</td>
<td>EC135 T3 EC135 T2, EC135 P3, EC135 T2+</td>
<td>Y-cope for double load hatch on a EC135 Human External CargoRev. 2: Transfer of STC to Airwork &amp; Heliseilerei GmbH</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements the following paragraphs at a later amendment: CS 27 Amendment 3:27.27; 27.301; 27.305; 27.309; 27.321; 27.337; 27.571; 27.603; 27.605.7.209; 27.610; 27.613; 27.625;...</td>
<td>- MDL-201212573 Rev NC, dated 15/08/2017. - Declaration of Compliance DOC-201212573 Rev NC, dated 01/12/2017. - ICA-201212573 Rev NC, dated 29/06/2017.</td>
<td>Applicable to EC135 helicopters with Dual External Cargo Hook, and related FAA Approval Number R.C.0167, and related FMS 9.2-47 Prior to installation of this design change it must be determined that the integration between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the other products.</td>
<td>19/03/2019</td>
<td>Active</td>
<td>10020379</td>
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<tr>
<td>1006383</td>
<td>AVONICS SUPPORT GROUP, Inc.</td>
<td>FAA STC ST04229AT</td>
<td>737-800/-800 737-300-400/500-600/700-800/900 series aircraft.</td>
<td>Installation of an International Communications Group (ICG) IridiumSatellite Communications System on B737-300/-400/500/600/700-800/900 series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List 214184-M, Revision C, dated 08 October/2015 or later FAA approved Revision.FAA approved Airplane Flight Manual Supplement for Avionics Support Group 214194-FMS dated 13 October 2015 or later approved FMS Instruction for Continued Airworthiness Document 214194-ICA Revision B.</td>
<td>Prior to installation of this design change it must be determined that the integration between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the other products.</td>
<td>07/12/2017</td>
<td>Active</td>
<td>10020362</td>
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<tr>
<td>1006391</td>
<td>BOMBARDIER Inc.</td>
<td>REV. 1</td>
<td>TCRA STC SA17-80 CL-600-2B10 CL-600-2B16 (604 VARIANT) CL-600-2B16 (601-3R VARIANT) CL-600-2B16 (601-3A VARIANT) CL-600-2A12 (601 VARIANT) CL-600-1A11 (600)</td>
<td>EASA validation of Transport Canada STC SA17-80 at Issue 2Modification to Water Line Heater Wiring Modification to be completed in accordance with B/E Aerospace LM Modification Data Summary Number L44000 Revision A or later approved/revision.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Modification to be completed in accordance with B/E Aerospace LM Modification Data Summary Number L44000 Revision A or later approved/revision.</td>
<td>Master Drawing List 214184-M, Revision C, dated 08 October/2015 or later FAA approved Revision.FAA approved Airplane Flight Manual Supplement for Avionics Support Group 214194-FMS dated 13 October 2015 or later approved FMS Instruction for Continued Airworthiness Document 214194-ICA Revision B.</td>
<td>Prior to installation of this design change it must be determined that the integration between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the other products.</td>
<td>15/02/2018</td>
<td>Active</td>
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<tr>
<td>1006398</td>
<td>STC TWENTY ONE LIMITED</td>
<td>FAA/UK RESPONSIBILITIES</td>
<td>777-200</td>
<td>Gogo 2KU - Structural Provisions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AMOC to FAA Airworthiness Directive 2012-07-06/0&lt;+&gt;/&lt;+&lt;+&lt; + later revisions of the above listed documents approved by EASA in accordance with...</td>
<td>This STC only approves the installation of the structural provisions for the installation of the antenna and radar. The installation of the antenna and radar and the loads they induce in the...</td>
<td>08/12/2017</td>
<td>Active</td>
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<td>1006400</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC SA18948W</td>
<td>SEE FAA AML</td>
<td>Modification of a Rockwell Collins Global Positioning System (GPS)/installationRev. 1: Update of GPS-4000S to resolve software issues Rev. 2: Correction of &quot;Limitations/Conditions&quot;section</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval, except where amended by additional or other amendments if indicated on FAA STC SA18948W. The requirements for environmental protection and the associated/required noise and/or emissions levels of...</td>
<td>(1) AML RCA-0392, Revision (-), dated 27 September 2017 (2) Instructions for Continued Airworthiness (ICA) RCA-0394, Revision(-) as specified by AML RCA-0392, Revision(-) or later FAA approved/accepted revision for items (1) and (2).</td>
<td>(1) Prior FAA approved installation of a Rockwell Collins GPS 4000S, P/N: 822-2189-001(003)-003(004)-05/06/07-0110-0111-100, is approved for this STC(2) The STC is valid for those aircraft types and models approved by EASA within the FAA AML RCA-0392, Revision(-), dated 27 September 201...</td>
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<td>10064001</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC 101833W</td>
<td>SEE FAA AML</td>
<td>Modification of a Rockwell Collins Global Positioning System (GPS) installation/Revision 1 updates the GPS-4000S.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC 101833W. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>(1) AML: RCAA-0324, Revision (-) dated 01 March 2017 (2) Instructions for Continued Amortkniness (ICA): RCA-0191, Revision (-) as specified by AML: RCAA-0324, Revision (-); or later FAA approved accepted revision for items (1) and (2).</td>
<td>08/02/2020</td>
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<td>10064008</td>
<td>PMV ENGINEERING</td>
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<td>A321-231/-233/25/261N</td>
<td>A321-111/-112/-113/-114/-115</td>
<td>ADBS-OUE installation on A320 family. The change consists of Transponder replacement or upgrade for DO-260B.Compliance of the installation of a dedicated ADS-B fail annunciator, seeing between the GNS5 receiver and the Transponder, and a switching between the Transponder and the Annunciator.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Document List AVI-01430-MDL-R01, dated 11 Dec. 2017 Aircraft Flight Manual Supplement AVI-01430-AFS-R00, dated 04 Aug. 2017 Instructions for Continued Amortkniness AVI-01430-ICA-R01, dated 08 Oct. 2017 For later revisions of the above listed documents approved by EASA.</td>
<td>11/12/2017</td>
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<tr>
<td>10064009</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
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<td></td>
<td>P.68R &quot;VICTOR&quot;</td>
<td>STC-AT-115 Additional NPAS Modifications</td>
<td>Details of the installed mission equipment can be found in the AFM Doc No STC-115-6-AFM</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative environmental protection requirements: Special Condition (s): CRI F-01, Lithium battery installations (rechargeable). The following paragraphs of CS 23: CRI: CS 23.01, CS 23.03, CS 23.305, CS 23.</td>
<td>Change Doctrine, Doc. No. STC-AT-115, Rev. 2 (1) Aircraft Flight Manual, Doc. No. STC-115-6-AFM, Rev. 1 (2) Aircraft Maintenance Manual, Doc. No. STC-115-6-AMM, Rev. 1 (3) Aircraft Maintenance Manual, Rev. 2 For later revisions of the above listed documents approved by EASA.</td>
<td>The following changes must be accomplished on the affected aircraft prior to the incorporation of STC-AT-115. EO-AT-095 Aircraft Preparation -- Airborne Technologies GmbH EASA Approval No. 10055442 Rep. BEO-AT-096 NPAS Mission Equipment Installation -- Airborne Technologies GmbH EASA Approval No. 10043960...</td>
<td>12/12/2017</td>
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<td>10064010</td>
<td>KLM ENGINEERING &amp; MAINTENANCE</td>
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<td>A330-300</td>
<td>Cabin Recconfiguration of KLM A330-300(Replacement of Business Class seats, installation of monuments, modification of In-Flight Entertainment System, replacement of deodorizers, curtains and carpet and related changes to cabin furniture to accommodate new Business Class cabin layout.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Data List, KLM Report MLA-03437, Revision 02, dated 7 Dec. 2017 - Instructions for Continued Amortkniness KLM Report ICA-333-13437 Revision ORG, dated 25 Oct 2017 For later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the ammortkniness of the product.</td>
<td>12/12/2017</td>
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<td>10064011</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>REV. 3</td>
<td></td>
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<td>SATCOM Astar 350 installation and mechanical installation of RGA-5017 antenna</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval...</td>
<td>Rev. 1: 66436-HDK01 Flight Manual Supplement Edition 2 For later revisions of the above listed documents approved by EASA...</td>
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<td>JET AVIATION AG</td>
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<td>Wireless Data Loader Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements... Special Condition (s): CRI F-31 Security Protection of Aircraft and Networks. The following paragraph(s) is/are later amendment: CS-25.301, 307, 561.789, 831, 869, 898, 1...</td>
<td>Master Data List: JBSL-M-00889ML-01 Issue 02 Certification Compliance Sheet: JBSL-M-00889CS-02 Issue 02</td>
<td>12/12/2017</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 3</td>
<td>FAA STC 704332NY</td>
<td>777-300ER, 777-200LR</td>
<td>Installation of an InletSat 2Ku Cabin Connectivity SystemRev. 1: Added 777-200-200LR modelsRev. 2: Amendment to the installation of a Doga 2Ku Cabin System on the Boeing 777-200-200LR-300ER Rev. 3: Amendment updates the 2Ku system from the utility Bus to the Main 115AC bus</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements Special Condition (s): CRF-102 Non-rechargeable Lithium Battery/Installations The following paragraph(s) is/are later amendment: CS-25.853 and CS-25.897sat Amendment 47Rev req...</td>
<td>- Master Data List 0143-1005-243s Revision AA dated 9 August 2001 or later FAA approved revisions - Airplane Flight Manual Supplement Ref. 043-1004-2439 Rev. -... dated 8 October 2017 or later FAA approved revisions - Instructions for Continued Airworthiness Ref. 0743-1004-2435, Rev C dated 29 Jun...</td>
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<td>TELEDYNE BATTERY PRODUCTS</td>
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<td>FAA STC SR06831DE</td>
<td>407</td>
<td>Installation of Teledyne Battery Products Gill LT Valve-Regulated Lead-Acid (L-T-VRLA), PIN 7407-28, battery</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>Teledyne Battery Products Document No. 7034-8-20, Master Data List Installation of Teledyne Battery Products PIN 7407-28, Revision IR dated 8 October 2016, FAA approved 1 December 2016 or later FAA approved revisions/Instructions for Continued Airworthiness (CA) Document No. 7034-6-03, Revision NC da...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>14/12/2017</td>
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<td>TELEDYNE BATTERY PRODUCTS</td>
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<td>FAA STC SA00757DE</td>
<td>525, 525A, 528B, 528C</td>
<td>Installation of Teledyne Battery Products PIN 7638-44 Lead-Acid Battery</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>Document No. 7034-4-02, Master Drawing List: Revision IR dated 17 May 2016 or later FAA approved revisions/Instructions for Continued Airworthiness (CA) Document No. 7034-6-01, Revision A dated 7 December 2015, FAA accepted 7 December 2015&lt;...or later FAA accepted revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>14/12/2017</td>
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<td>TELEDYNE BATTERY PRODUCTS</td>
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<td>FAA STC SA0687DE</td>
<td>PC-12/47E, PC-12/47F</td>
<td>Installation of Teledyne Battery Products Valve-Regulated Lead-Acid(VRLA), PIN 7638-44, battery</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>Teledyne Battery Products Document No. 7074-1-01, Master Drawing List:Installation, Revision IR dated 24 June 2017, FAA approved 30 June 2017 or later FAA approved revisions/Instructions for Continued Airworthiness (CA) Document No. TD 74-8-02, Revision IR dated 22 June...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>14/12/2017</td>
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<td>List of EASA Supplemental Type Certificates</td>
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<td><strong>Rev.</strong></td>
<td><strong>Foreign/NAA STC No.</strong></td>
<td><strong>Model</strong></td>
<td><strong>Description</strong></td>
<td><strong>Certification Basis</strong></td>
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<td>10064056</td>
<td>IKHANA GROUP, LLC</td>
<td>REV. 2</td>
<td>FAA STC SA0346LA</td>
<td>DHC-6 SERIES 400 DHC-6 SERIES 300 DHC-6 SERIES 200 DHC-6 SERIES 100 DHC-6-100/200/300-400</td>
<td>Installation of Max-Vs 2300 Enhanced Vision SystemRev. 1: Name Change from &quot;Ikhana Group Inc.&quot; to &quot;Ikhana Group LLC.&quot;</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unamended and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>FAA approved revision</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/adverse effect upon the airworthiness of the product.</td>
<td>13/06/2022</td>
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<td>10064057</td>
<td>AEROCLUBUL ROMANIEI</td>
<td>IS-28B2</td>
<td>IS 28B2: TBO increase Increasing TBO from 1200 flight hours / 7500 landings to 1400 flight hours / 8400 landings in accordance with Aeroclubul României document IS-28B2-AR-03, at Revision 2, dated 25 October 2017.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unamended and remain applicable to this certificate/approval.</td>
<td>- Flight and Maintenance Manual edition 2A, dated 1977, pages 30, 32, 33, 26, 28, 1, 36, 37, - Flight and Maintenance Manual edition 3, dated 1978, pages at amendment no 44, 0.2, 0.3.2.1, 6.9, 6.9.1, 6.10, 6.14, 6.15 or later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/adverse effect upon the airworthiness of the product.</td>
<td>14/12/2017</td>
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<td>10064059</td>
<td>DIAGNOSTIC SOLUTIONS INTERNATIONAL</td>
<td>FAA STC SR02630LA</td>
<td>AS 350 B2 AS 350 B1</td>
<td>Installation of Honeywell Model 1517SVX Health Monitoring System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unamended and remain applicable to this certificate/approval.</td>
<td>Diagnostic Solutions International Master Drawing List, Document Number 25715, Rev. &quot;A&quot;, dated 15 December 2016, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/adverse effect upon the airworthiness of the product.</td>
<td>14/12/2017</td>
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<td>10084060</td>
<td>TELEDYNE BATTERY PRODUCTS</td>
<td>FAA BASIC STC ST0096DE</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of Teledyne Battery Products PIN 7638-447 Lead-Acid Battery</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unamended and remain applicable to this certificate/approval.</td>
<td>Document No. TD03-4-01, Master Drawing List, Revision IR dated 04 March/2016 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/adverse effect upon the airworthiness of the product.</td>
<td>14/12/2017</td>
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<td>1008470</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>PA-31T (CHEYENNE II), PA-31T2 (CHEYENNE III), PA-31T1 (CHEYENNE I), PA-31P PRESSURIZED NAVAJ, PA-31P-350 (MOJAVE)</td>
<td>Installation of one or more of the following antennas: Single Germi OASIS GPS antenna - Single L-3 Communications NY-164 antenna.</td>
<td>The Certification Basis for the original product as amended with additional or alternative avionics requirements:The following paragraphs at a later amendment: 23.29, 23.30(a), 23.30(b), 23.30(d), 23.30(e), 23.30(f), 23.30(g), 23.30(h), 23.30(i), 23.30(j), 23.37, 23.37(a), 23.37(b), 23.37(c), 23.37(d), 23.37(e), 23.51(a), 23.51(b), 23.51(c), 23.51(d), 23.51(e), 23.51(f), 23.51(g).</td>
<td>- Master Documents List ASR-2017-055-MDL-01-00, dated 22/11/17; - Instructions for installation ASR-2017-055-P1-00, dated 22/11/17;</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/adverse effect upon the airworthiness of the product. The STC is valid for those aircraft types and model specific ICAO documents.</td>
<td>15/12/2017</td>
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<td>10084703</td>
<td>airbus engineering GmbH</td>
<td>REV. 2</td>
<td>FAA STC ST0096DE</td>
<td>Installation of Diversity Antenna.</td>
<td>The Certification Basis for the original product as amended with additional or alternative avionics requirements - The following paragraphs at a later amendment: CS-ACNS.D.ELS, 060-CS-ACNH.D.ELS, 065. The requirements for environmental protection and the associated certificated noise and...</td>
<td>- Master Document List ASR-2017-055-MDL-01-00, Issue 3; - Description of Modification DE-2017-055, Issue 3; - Means of Compliance List ASR-2017-055-MDL-01-00, Issue 3; - Maintenance Manual Supplement MM-2099-00000, Issue 3;</td>
<td>This modification is only applicable to aircraft, which have been modified with the DLR Lightning conversion STC (EASA STCNo. 100016000) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change...</td>
<td>10/02/2021</td>
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### List of EASA Supplemental Type Certificates

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<th>Rev. level</th>
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<th>Certification basis</th>
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<td>10064801</td>
<td>DIAGNOSTIC SOLUTIONS INTERNATIONAL</td>
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<td>FAA STC SR02530LA</td>
<td>MBB-BK 117 C-2 MBB-BK 117 C-1</td>
<td>Installation of Honeywell Model 15175 VOP Health Monitoring System on Airbus Helicopters Deutschland, Model MBB-BK 117 C-1 and MBB-BK 117 C-2</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA approved Master Drawing List, Document Number 25989, Revision &quot;NC&quot; dated 20 March 2015, or later FAA approved revisions.</td>
<td>15/12/2017</td>
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<td>10064081</td>
<td>TELEDYNE BATTERY PRODUCTS</td>
<td>REV 1</td>
<td>FAA STC SR08642DE</td>
<td>EC135 T1, EC135 T2, EC135 T2+ EC135 P1, EC135 P2, EC135 P2+</td>
<td>Installation of Teledyne Battery Products Gill LT Valve-Regulated Lead-Acid (LT-VRLA) PIN 7638-44, battery</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Teledyne Battery Products Document No. TD70-4-03, Master Drawing List, Installation of Teledyne Battery Products PIN 7638-44, Revision IR dated 4 March 2017, FAA approved 7 March 2017 or later FAA approved revision. Instructions for Continued Airworthiness (CA), Document No. TD 76-8-C2 Revision NC dated 4 March 2017.</td>
<td>19/12/2017</td>
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<td>10064109</td>
<td>P3 ENGINEERING GMBH</td>
<td></td>
<td>767-300</td>
<td>767-300</td>
<td>First Class Reconfiguration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Declaration of Compliance of P3 Engineering GmbH Ref. P3-JAB01-DCG-0001Issue A. Certification Programme Compliance Summary Ref. P3-JAB01-CPH-0001 Issue C dated 14 December 2017-Master Document List Ref. P3-JAB01-MLD-0001 Issue A-Dated 15 December 2017. Prior later revisions of the above listed documents.</td>
<td>None</td>
<td>18/12/2017</td>
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<td>SABENA TECHNICS BOD</td>
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<td>A330-323</td>
<td>French Blue – G12 Ultra High Comfort Attendant Seats (UHCAS) installation. This modification installs 1 new UHCAS on the galley G12. and arranges the area by re-routing curtains, adding placards and relocating emergency equipment.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Master Data List, ref MDL-S-26-4128, Issue A, dated 17 November 2017.</td>
<td>None</td>
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<td>GOODRICH CARGO SYSTEMS (UNITED TECHNOLOGIES AEROSPACE SYST</td>
<td>REV. 1</td>
<td>FAA STC ST02546CH</td>
<td>10064118</td>
<td>For all models except 747-400F Install Power Drive Unit for Main Deck Cargo Handling Systems. For 747-400F Install Goodrich Freighter Common Control/Installations (FCCI) for Main Deck Cargo Handling Systems.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated notified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Summary Report ref. AS/AJA-17073 Issue 4 (Initial), AS/AJA-18077 Issue 4 (Major Changes), Modification Approval Sheet ref. MAS CJ2153 Issue 3: Maintenance Planning Document ref. MPS-5 MOD CJ22501, dated 01 December 2017, or later revisions of the above listed documents approved.</td>
<td>18/12/2017</td>
<td>Active</td>
<td>10064118</td>
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<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>REV. 2</td>
<td>EC136 T2+, EC136 T3</td>
<td>10064126</td>
<td>INSTALLATION OF FLIGHTCELL DZMX WITH OPTIONS FOR WIT.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated notified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- AFM Supplement 2109331-AM, Issue 1&lt;-1&gt; &lt;-1&gt; =Engineering Order &amp; Certification Program 2109331-EO, Issue 1 Certification Documents 2109331-DOC, Issue 1 or later revisions of the above listed documents approved for change or repair.</td>
<td>27/07/2020</td>
<td>Active</td>
<td>10064126</td>
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<tr>
<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>REV. 1</td>
<td>DORNIER 228-212</td>
<td>10064134</td>
<td>Aerostretch®R System Installation for Dornier 228 Series Aircraft. In accordance with GvH Aerospace modification ref GvH-7528-01-MOD-01, Issue 1 dated 13.12.2017 Rev. 1. Transfer of STC from GvH Aerospace Ltd - United Kingdom, to GvH Aerospace (Ireland) Limited.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated notified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with GvH Aerospace modification ref GvH-7528-01-MOD-01 and Master Document List GvH-7528-01-MDL-01, Issue 1 dated 12.12.2017.Operation in accordance with GvH Aerospace Pilot’s Operating Handbook/Supplement ref GvH-7528-01-FMS-01, Issue 2, dated 15.11.2017.Instructions for Co...</td>
<td>26/01/2021</td>
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<td>EASA STC no.</td>
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<td>10064136</td>
<td>ASI INNOVATION</td>
<td>1</td>
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<td>EC135 T2+</td>
<td>Installation of multipurpose container POD on EC135 multi-function armament.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplement Manual de Vol installation de l’OD 5620 sur bras multifonction EC135 T2+ &amp; T2e&quot;, document ASI- AFM-9405, Rev. 11-12 or later revisions of the above listed documents approved by EASA - Certification Programmes ASI- CP-0406-01, Rev. 1 - SD- STC Data List ASI- SDL-3451-01, Rev. 4 - Master Drawing.</td>
<td>This STC approves the installation of the multipurpose container only. Any further installation in- or outside of the container requires further approval. The installation is limited to rotorcraft with &quot;Standard&quot; landing gear. Prior to installation of this design change it must be determined that the airworthiness of the product.</td>
<td>19/12/2017</td>
<td>Active</td>
<td>10064136</td>
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<td>10064143</td>
<td>ARCHIEON HOLDINGS, L.L.C.</td>
<td>REV. 3</td>
<td>FAA STC ST04215AT</td>
<td>737-800/-900/-700</td>
<td>Installation of Avionics sat LINK MAX with optional Personality Module (PM) in the Cabin Compartiment overhead, and optional vRDC MAX QAR in the EE Bay. Existing aircraft systems optional Interface/Integration include the Flight Deck Audio System (i.e. Pilot Crew Headsets), MCDU’s (or IDUs), ACARS CMU &amp;...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List 142335- MDL, Revision 1 dated 19 September 2018.</td>
<td>This approval should not be extended to other aircraft of this model unless previously approved modifications are incorporated. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/02/2019</td>
<td>Active</td>
<td>10064143</td>
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<tr>
<td>10064146</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 2</td>
<td>A320-271N/-271NX/-272N</td>
<td>CABIN AIR RECIRCULATION FILTER CARTRIDGE REPLACEMENT ON AIRBUS SINGLE AILSELT. Rev. 2 - Correction of Applicability list.</td>
<td>The Type and OSO Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>- Instructions for Continued Amorthelessness, AEROCONSIL document No. 784-01-A-MDL, Rev 05, dated 15 November 2002; latest revisions of the above listed document(s) approved/superseded by the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the inter relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/11/2022</td>
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<td>10064158</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA ST01785WI</td>
<td>MYSTERE-FALCON 900</td>
<td>FAA ST01785WI Installation of a Data Link Communication System supporting Future AirNavigation System (FANS 1A+)</td>
<td>Validation of FAA STC ST01785WI Installation of a Data Link Communication System supporting Future AirNavigation System (FANS 1A+).</td>
<td>- Master Document List No. EDOC-018246, Revision B, dated 24 September 2015+ (&lt;,&gt;) or later revisions of the above listed documents approved by EASA - Certification Programmes ASI- CP-0406-01, Rev. 1 - SD- STC Data List ASI- SDL-3451-01, Rev. 4 - Master Drawing.</td>
<td>- This STC is applicable to Dassault Aviation Mystère- Falcon 900 airplanes S/N 001 through 178. - Installation of this STC requires installation of EASA STC 10060212 and approved Satellites. Communication system that meets the required performance on 20-140B. Prior to installation of this design change it must...</td>
<td>20/12/2017</td>
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<td>STC Holder</td>
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<tr>
<td>AIRBUS DEFENCE AND SPACE S.A</td>
<td>A330-243</td>
<td>Conversion of A330-200 Series A/C to A330 Multi Role Transport Tanker (MRTT) F/AF (French Air Force). Revision 1 of this STC, approval of the ETOPS capability of the basic aircraft to the MRTT FAF aircraft. Revision 2 of this approval introduces AIS Part 2 Supplements for MRTT's STCs based on A330 ALI Issue...</td>
<td>- CONFORMANCE DEFINITION A330 MRTT FAF STC document Ref. TAE-FF-FT-170082 Issue A. Airline Flight Manual Supplement “EASA APPROVED MODEL A-330 MRTT FAF/AIRLINE FLIGHT MANUAL” document Ref. TAE-FF- DT-150509. Issue associated to STC 10069541 AML TEA-FA-AIR-100/195. Artico Defence and Space, Tempora...</td>
<td>Approval of the STC is limited to compliance to selected Certification Standards as defined in A330 MRTT FAF CRJ A-01/A330 MRTT RAAP CRI G-01. It defines in particular restrictions related to Maintenance Tasks and Airworthiness Limitations, Ref. JAR 25.57143 and or/and 5182. Therefore, the following limitations apply...</td>
<td>17/04/2023</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>GV, GV-SP</td>
<td>Civil Aircraft Missile Protection System (CAMPS) Installation.</td>
<td>The Certification Basis for the original product as amended by the above-listed additional or alternative airworthiness requirements - Special Condition (s): D-200304-02. Partial Empty Cabin, in interest of Governmental Service. D-200304-02. Air Medical Service. The following paragraph(s) at a later amendment: CS 25 Amendment 2: The requirements affected by this change. The requirements...</td>
<td>- Aircraft Flight Manual, Doc Ref: 2105396-AM-2, Continuation/airworthiness for CAMPS Installation, Doc Ref: 2105396-CA.Ca later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/12/2017</td>
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<td>TURK HAVA YOLLARI TEKNIK A.S (TURKISH TECHNIC INC.)</td>
<td>737-800</td>
<td>Cabin Reconfiguration with introduction of New Q1, G2, Q4 Galley inserts</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>CO020-CEP-04 Certification Programme, dated 21/12/2017</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/12/2017</td>
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<td>10064207</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC STD/2017/300ER</td>
<td>Installation of a go2 KRU Radome (with Doppler opt/def). 1. Added model 777-200 Rev 2: Update of Master Data List. Revision 4. Update of Master Data List revision and Structure...</td>
<td>The Certification Basis for the original product as amended by the above-listed additional or alternative airworthiness requirements - Special Condition (s): Refer to CS-F-01. The following paragraph(s) at a later amendment: CS 25 Amendment 2: The requirements affected by this change. The requirements...</td>
<td>Master Data List Ref. 0143-10287-2434 Rev. AR, dated 8 May 2023&gt;(,&lt;) &gt;Airline Flight Manual Supplement Ref. 0403-10286-2434 Rev. B,dated 20/04/2018&gt;(,&lt;) &gt;Instructions for Continued Airworthiness Ref. 0743-10288-2434 Rev. P, dated 2 June 2023&gt;(,&lt;) &gt;EWS Instructions for Continued Airworthiness Ref. 0743-10288-2434 and EWS ICA Ref. 0743-10288-2434 Prior to installation...</td>
<td>This STC includes the validation of the Alternative Method of Compliance FAA Ref. 792.17-2796 or 782.18-682) to AD 2012-07-06 AD 2013-07-11. The customer must be accomplished prior to installation. For additional limitations refer to the ICA Ref. 0743-10288-2434 and EWS ICA Ref. 0743-10288-2434 Prior to installation...</td>
<td>06/06/2023</td>
<td>Active</td>
<td>10064210</td>
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<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>P2006T</td>
<td>Aeronautical installation of the mission system Leica City Mapper (Sensor Unit) and respective server, cockpit display and operator stand and display in Tecnam P2006T aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>-P2006T10-HDK01 Issue 4 FMS MISSION EQUIPMENT IN P2006T/ In a Cessna Citation Mustang, ..</td>
<td>The following OEM modification must be implemented in the aircraft prior to carry out this design change...</td>
<td>22/12/2017</td>
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<td>10064224</td>
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<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>DUAL TRANSPLANTER SYSTEM ADS-B UPGRADE IN PART 26 AIRCRAFT (VARIOUS)</td>
<td>The Certification Basis for the original product as amended by the above-listed additional or alternative airworthiness requirements - Special Condition (s): CS 25 amendment 2 for the affected requirements. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product...</td>
<td>-Airline Flight Manual, Doc Ref: 210596-AM A-Aircraft Flight Manual, Doc Ref: 210596-AM B-Continuied Airworthiness, Doc Ref: 210596-CA A later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/12/2017</td>
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<td>10064226</td>
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<td>STC Holders</td>
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<td>Rev. level</td>
<td>FAA STC</td>
<td>STC ID (if)</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STC</td>
<td>ST025005SE</td>
<td>787-9</td>
<td>First class seat upgrade on Boeing Model 787-9 series airplanes.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. B/E Aerospace, Inc. - FSI (B/E - FSI) Master Drawing List (MDL) BIFS707-001, Revision B, dated 15 November 2017 or later Federal Aviation Administration (FAA) approved revision. 2. The Supplemental Instructions for Continued Airworthiness, B/E - FSI Document BIFS707-015, Revision B, dated 14 November...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>01/01/2018</td>
<td>Active</td>
<td>10064251</td>
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<td>KNSI AVIATION ENGINEERING LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A330-343</td>
<td>Flush doubler installation for Antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>KNSI Instructions for continued airworthiness 17K024- ICA-001-R Revision 03 dated 13 December 2017 (&gt; =)or later revisions of the above listed documents approved by EASA-KNSI Master Document List 1PK024-MDL-001-5-R. Revision 05 dated 15 December 2017.</td>
<td>This modification de-modifies GOOD 2KU antenna systems installed in accordance with EASA STC 10058732. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/01/2018</td>
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<td>AEROCET, INC.</td>
<td>FAA STC</td>
<td>SA12575SE</td>
<td>Installation of Aerocet Model 3400 floats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Aerocet Drawing List Drawn on FAF AML, SA012575SE, dated 28 August 2001 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/01/2018</td>
<td>Active</td>
<td>10064257</td>
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<td>ANDERSON, THOMAS RICHARD</td>
<td>FAA STC</td>
<td>SA01875SE</td>
<td>Installation of titanium main landing gear assembly</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- FAA approved Master Drawing List XP010101MDL, Revision D dated 14 August 2014</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/01/2018</td>
<td>Active</td>
<td>10064258</td>
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<td>AVIATION TECHNICAL SERVICES, Inc.</td>
<td>FAA STC</td>
<td>ST02575SE</td>
<td>Installation of vortex generators</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. Aviation Technical Services Master Data List (MDL) 780001-MDL-001-R Revision A, dated 12 October 2017, or later Federal Aviation Administration (FAA) approved revision. 2. Instructions for Continued Airworthiness (ICA) 780001-ICA-001-R Revision A, dated 7 November 2017, or later FAA accepted revision...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/01/2018</td>
<td>Active</td>
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**List of EASA Supplemental Type Certificates**

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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<tr>
<td>10084260</td>
<td>KNSI AVIATION ENGINEERING LIMITED</td>
<td>CA An RESPONSIBIL ITY</td>
<td>A340-642</td>
<td></td>
<td></td>
<td>Flush doubler installation for Antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. ...</td>
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<tr>
<td>10084261</td>
<td>FTI ENGINEERING NETWORK GmbH</td>
<td>FAA STC ST0367AT</td>
<td>525, 525A, 526B</td>
<td></td>
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<td>ADS-B Out for Citation Jet.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-25: Am4, 23, 23.123.21, 23.20, 23.601, 23.603, 23.605, 23.607, 23.1301, 23.1305, 23.1322, 23.1351, 23.1357, 23.13...</td>
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<td>10084263</td>
<td>ARCHEION HOLDINGS, L.L.C.</td>
<td>FAA STC ST0367AT</td>
<td>525, 525A, 526B</td>
<td></td>
<td></td>
<td>Installation of an Avionics, Inc. aWiFi wireless router system. The system provides 3 modes of operation: laptop in user position, device to device Wi-Fi wireless, or Wi-Fi connectivity to the Ground Terminal/Wi-Fi Router on the ground v...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis for Continued Airworthiness 170206- ICA-001-3.R Revision 03 dated 13 December 2017 &lt;(&gt;) later revisions of the above listed documents approved by EASA.</td>
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<td>10084268</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>FAA STC ST0367AT</td>
<td>525</td>
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<td>GA35 GPS ANTENNA INSTALLATION in accordance with Scandinavian Avionics Engineering Order 2106348-EO</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis for Continued Airworthiness 170206- ICA-001-3.R Revision 03 dated 13 December 2017 &lt;(&gt;) later revisions of the above listed documents approved by EASA.</td>
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<td>10084269</td>
<td>FOKKER SERVICES B.V.</td>
<td>FAA STC ST0367AT</td>
<td>A321-253N, A321-271N, A321-231, A321-233, A321-235N, A321-237N</td>
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<td>Introduction of ADS-B OUT capabilities by addition of DO-260B capable ATC transponder.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis for Continued Airworthiness 170206- ICA-001-3.R Revision 03 dated 13 December 2017 &lt;(&gt;) later revisions of the above listed documents approved by EASA.</td>
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This modification de-modifies GOOD 2KU antenna systems installed in accordance with EASA STC 10058069. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.
10064282 AVODYNE CORPORATION REV. 1 FAA STC SR003780C R66 R44, R44 II R22, R22 ALFA R22 BETA, R22 MARINER Installation of Avodyne 700-00182-XXX (FDS) and 700-00179-XXX (FD4) Series Integrated Flight Displays (IFD) The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements or regulations at the latest amendment 23-0 through 27-4727.1(a)(b), 27... 1. Avodyne Corporation Master Document List Document Number VFD-4 95, Revision 02, dated 6 December 2017, or later FAA approved revision.2. Avodyne Corporation Rotorcraft Flight Manual Supplement (RFMS) Document Number 600-00324-000, Revision 01, FAA approved on 12 December 2017, or later FAA approved. This STC only covers the installation of the IFDs, the IFD bays, and associated wiring. It does not cover the installation of antennas, annunciators, or switches that may be required to make a fully functional system. Prior to installation of this design change it must be determined that the interference between the design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. STC is valid for those aircraft types and makes.

10064296 CERT WORKS, LLC REV. 1 FAA STC SR06812DE SEE FAA AML Daily Use of Honeywell Sky Connect Tracker III System Rev.1: Addition of Bell 429 to the FAA Approved Model List (AML) and update of document numbers. The certification basis for parts changed or affected by the change involving the reference date of application is: <= March 16, 2016, is based upon 14 CFR Part 27, Amendment 27-1 through 27-14 CFR Sections: 27.301; 27.303; 27.305; 27.307(a); 27.307(a) Amdt. 27-26,... Document No. CW9063-1006, Master Installation Drawing List Part 27 AML Sky Connect Tracker III Installation, Revision 01, dated 29 August 2016 or later FAA approved revisions. Parts manufactured in accordance with Document No. CW9064-1004, Master Installation Drawing List, Document No. CW9063-1006, Part(s)... Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. STC is valid for those aircraft types and makes. STC is valid for those aircraft types and makes.

10064297 BOMBARDIER Inc. TCCA SA17-93 CL-600-2B16 (SPE. ED. SE ONLY) CL-600-2B16 (604 VARIANT) CL-600-2B16 (601-3R VARIANT) CL-600-2B16 (601-3A VARIANT) CL-600-2A12 (601 VARIANT) CL-600-1A11 (600) EASA Validation of Transport Canada STC SA17-93 at Issue 1: Portable Oxygen Bottle Bracket Replacement. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Modification to be completed in accordance with See Airlora Document Ltd. Modification Data Summary Document Number L585000 Revision NC, dated September 22nd, 2017 or later approved revisions. Maintenance to be completed in accordance with Bombardier Instruction for Continued Airworthiness BAS63058026, Basic Issue, da... See those of original Transport Canada STC SA17-93. Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10064301 STARLING AEROSPACE LIMITED FAA STC SA04053CH 737-500 New Cabin Configuration. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: 23-26 Amendment 18; CS-25.23, CS-25.25, CS-25.27, CS-25.29, CS-25.301-CS-25.32, CS-25.30, CS-25.305, CS-25.307, CS-25.3561,... Starling Aerospace Limited Master Document List No. B737-MLD-0022 Issue 4. Restricted to MSN 24919. STC is valid for those aircraft types and makes.

10064303 WIPARE Inc. FAA STC SA04053CH T206H 18SE, A18SE, A18F 18S, 18SA, 18SB, 18SC, SE Installed of Hartzell "Trailblazer" Composite Propeller in accordance with Wipare Inc. Installation Drawing as listed on AML No. SA04053CH issued November 23, 2016, or later FAA approved. -Wipare Inc. Installation Drawing 1008739, Revision 8, issued 12/15/15-Wipare Inc. Installation Drawing 1008739, Revision 8, issued 10/17/15-Wipare Inc. POHSA04053CH-1, Revision IR, issued 12/23/15-Wipare Inc. POHSA04053CH-3, Revision IR, issued 11/21/16or later revisions of the above listed... The installation for models 185/ 185B/D/E: A18SEF is limited to configurations with floats only. Prior to installation of this design change it must be determined that the interference between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.
### List of EASA Supplemental Type Certificates

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<td>HOWEYLY DIVISIONE SAM S.R.L</td>
<td>AS 350 B3 FreeFlight RA-4000 Radar Alt. Install.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-27 Amendment 4 date 30.11.2016 §27.21, 25, 27, 29, 45, §27.301, 303, 305, 321, 337, 351, 561, 571 §27.601, 603, 605, 609...</td>
<td>FAA STC no. STC:CS 29 Amendment 3:CS 29,45 §27.601, 303, 321, 337, 351, 561, 571 §27.601, 603, 605, 609...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>12/01/2018</td>
<td>Active</td>
<td>10064307</td>
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<tr>
<td>BOMBARDIER Inc.</td>
<td>STC SA17-83 BD-700-1A11 BD-700-1A10 Installation, Underwater Locator Device</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (a)-(F)-10052103-1 at issue 1 – Non-Rechargeable Lithium Battery Installations The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Approved Tyler Camera Systems Master Drawing List No. ML/0133-025, Issue 06, dated 20 January 2003 FAA Approved Rotordraft Flight Manual Supplement No. OR-RFM/001, Reissued 6 August 2004, or later FAA Approved Revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>12/01/2018</td>
<td>Active</td>
<td>10064315</td>
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<tr>
<td>FAABASIC STC SR01446A</td>
<td>Installation of cargo rack.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Approved Tyler Camera Systems Master Drawing List No. ML/0133-025, Issue 06, dated 20 January 2003 FAA Approved Rotordraft Flight Manual Supplement No. OR-RFM/001, Reissued 6 August 2004, or later FAA Approved Revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>15/01/2018</td>
<td>Active</td>
<td>10064325</td>
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<tr>
<td>REV. 2 BACCOCK CRITICAL SERVICESDESIGN N AND COMPLETIONS LIMITED</td>
<td>MBB-BK117 D-2 Wales Air Ambulance mission configuration including NVIS, ref BD/133-100-01 Rev. 1 introduces an NVIS-certified right-hand configuration option for the A800 Trakka searchlight and additional LED Lading Lights mounted online and aft skid cross tubes (validation on CAA UK STC UK-STC.00070Rev. 0) Rev. 2...</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: For the original STC CS 29 Amendment 3 CS 29.1 (a), CS 29.21(a), CS 29.29 (a) (b), CS 29.49 (a) (1) (3), CS 29.51 (a) (1)...</td>
<td>FAA Approved Tyler Camera Systems Master Drawing List No. ML/0133-025, Issue 06, dated 20 January 2003 FAA Approved Rotordraft Flight Manual Supplement No. OR-RFM/001, Reissued 6 August 2004, or later FAA Approved Revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>30/06/2021</td>
<td>Active</td>
<td>10064359</td>
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<tr>
<td>FAASERVICE COMPANY</td>
<td>STT10952SC 737-700 Installation of an active ventilation system for lavatories, galleys and non-essential equipment throughout the aircraft</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: Part 25 of the Federal Aviation Regulations as amended by Amendments25-1 through 25-77 with the exceptions listed below (Am...</td>
<td>1. Single fan– Master Documents List (MDL) 08/10/2002 Revision A dated 9 September 2002, or later FAA approved revisions...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness of the product.</td>
<td>20/02/2018</td>
<td>Active</td>
<td>10064374</td>
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<td>STC Holder</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>Validation of FAA STC ST02974NY Installation of Data Link Communications System with Honeywell Primus2000 Avionics System.</td>
<td>ST02974NY</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s) CRI F-9122-001: Flight Recorders and Data Link Recording, including the following paragraph (s) at a later amendment: CS25 Amendment 18 of CS25-1301, CS 25-1457 a...</td>
<td>F9122-001</td>
<td>18/01/2018</td>
<td>Active</td>
<td>10043752</td>
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<tr>
<td>SEBALD LUFTFAHRTTECH NISCHER BETRIEB(LTB)</td>
<td>TM-S01 „Lenkbares Spornrad N4M“ / SB-S01 „Steerable tail wheel N4M“ Lenkbares Spornrad für Nimbus-4M / steerable tailwheel for Nimbus-4M</td>
<td>Nimbus-4M</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/repair. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/repair.</td>
<td>S01</td>
<td>19/01/2018</td>
<td>Active</td>
<td>10044412</td>
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<tr>
<td>VOYAGEUR AEROTECH, Inc.</td>
<td>DHC-8-102 DHC-8-103/ DHC-8-106</td>
<td>DHC-8-106</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/repair. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/repair.</td>
<td>DHC-8-106</td>
<td>19/01/2018</td>
<td>Active</td>
<td>10044410</td>
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<tr>
<td>365 AEROSPACE DDS LIMITED</td>
<td>New Cabin Configuration Rev. 1: Transfer of STC from 365 Aerospace Ltd. to 365 Aerospace DDS Ltd.</td>
<td>737-300</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25 Amendment 18:25.152(2)&lt; (&gt;) 25.1701(&lt;&gt;) &gt; 25.1715(&gt;) 25.1715(&gt;) 25.1715(&gt;) 25.1715(&gt;) 25.1715(&gt;) 25.1715(&gt;) 25.1715(&gt;) 25.1715(&gt;) &gt; 25.17...</td>
<td>365 Aerospace Flight Manual Supplement No. FMS-0277-001 Revision A365 Aerospace Instructions for Continued Airworthiness No. ICA-0277-018 Revision A44 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>05/02/2019</td>
<td>Active</td>
<td>10044419</td>
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<tr>
<td>AIRBUS S.A.S.</td>
<td>MOD CJ2096 - VASAT KA BAND Antenna Installation on Single A350XWB STC covers the installation of structural</td>
<td>A321-231</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/repair. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/repair.</td>
<td>CJ2096</td>
<td>None</td>
<td>None</td>
<td>10044419</td>
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<td>STC no.</td>
<td>STC Holder</td>
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<tr>
<td>10064446</td>
<td>JET AVIATION AG</td>
<td>Rev. 2</td>
<td>747-300, 747-400</td>
<td></td>
<td>KA Band Antenna Installation. The installation is composed of 3 configurations: CONF1: KA Band Structural Provision, CONF2: KA Band Antenna Systems Installation, CONF3: KA Band Antenna Installation (CONF1 and CONF2 combined). Revision 1 to extend the applicability to the 747-300 with a different antenna installation. The Certification Basis for the original product as amended (including additional or alternative airworthiness requirements) and/or noise and/or emissions levels is determined by the installer. The requirements for environmental protection and the associated certifications are applicable to this certificate/approval.</td>
<td>- JBL-M-0007MDL-01 Issue 05. JBL-M-0007MDL-01 Issue 06. JBL-M-0007MDL-01 Issue 07. JBL-M-0007MDL-01 Issue 08. JBL-M-0007MDL-01 Issue 09.</td>
<td>Prior to installation of this change/repair it must be determined whether this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10064445</td>
<td>Q.C.M. DESIGN GmbH</td>
<td></td>
<td>HAWKER 800XP</td>
<td></td>
<td>Delta Certification of a GPS/WAAS Sensor &amp; Antenna Installation for the support of the Approved EASA STC 10058529.</td>
<td>- Supplemental Type Definition Document, Ref. S6010-17/20 Issued Issue 01 Rev. 00.</td>
<td>-</td>
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<tr>
<td>10064466</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A340-641</td>
<td></td>
<td></td>
<td>RTAF Cabin Retrofit. The Certification Basis for the original product as amended (including additional or alternative airworthiness requirements) and/or noise and/or emissions levels is determined by the installer. The requirements for environmental protection are applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document, Ref. S6010-17/20 Issued Issue 01 Rev. 00.</td>
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<td>10064457</td>
<td>AERO SEATS AND SYSTEMS, INC</td>
<td>FAA STC</td>
<td>SR00340/17C</td>
<td>412</td>
<td>Installation of Aero Seats and Systems, Inc., Life Raft Deployment System (LRDS)</td>
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**List of EASA Supplemental Type Certificates**

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[Link to EASA website for the full list](easa.europa.eu)
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<td>32B SUPPORT SERVICES GmbH</td>
<td>STC</td>
<td>REV. 1</td>
<td>STC</td>
<td>10064476</td>
<td>FALCON 900EX (F900EX)</td>
<td>Installation of Garmin GRA 5500 Radar</td>
<td>- Master Document List No.MDL.118Z001, Revision B, dated 24th January 2018. - Airplane Flight Manual Supplement Document No. AFMS020ED3420001, Revision A, dated 15th January 2018. - Instructions for Continued Airworthiness Document No. JAR-MOS20ED3420001, Revision A, dated 18th January 2018 or later revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>24/01/2018</td>
<td>Active</td>
<td>10064476</td>
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<td>GARMIN DEUTSCHLAND GmbH</td>
<td>STC</td>
<td>REV. 1</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>10064473</td>
<td>Installation of Garmin GS Electronic Flight Instrument Rev.1 changes the STC holder from Garmin (Europe) Limited to GarminDeutschland GmbH.</td>
<td>The certification basis for parts not changed or not affected by this change remain unchanged from the original approval.</td>
<td>Installation in accordance with Garmin Master Drawing List (MDL) 005-0112-01, Revision 8, dated October 2018. Instructions for Continued Airworthiness (ICA) as listed in above MDL.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>12/02/2019</td>
<td>Active</td>
<td>10064473</td>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>STC</td>
<td>REV. 2</td>
<td>FAA SR96094RC</td>
<td>10064475</td>
<td>AS 350 B3</td>
<td>Installation of Garmin GRA 5500 Radar Attimeter System Rev.1, adds configuration 350A12-1656-02Rev.2: Correction of typos</td>
<td>- FAA approved by EASA November 13, 2017 or later</td>
<td>November 14, 2017 or later accepted revision.</td>
<td>For configuration 350A12-1656-01, this STC is applicable to AS350B3/Arcraft in the Post Mod AMS 07-4723 only. For configuration 350A12-1656-02, this STC is applicable to AS350B3/Arcraft in the Post Mod ECP 07-4061 only. Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>19/06/2023</td>
<td>Active</td>
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<tr>
<td>RAMI AEROSPACE INC.</td>
<td>STC</td>
<td></td>
<td>TCCA SH16-38</td>
<td>10064476</td>
<td>R44, R44 II</td>
<td>Installation of Replacement Floor Covering Replacement floor covering kits consisting of carpet, utility flooring and floor trays.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>MDL-RH544-1009 Master Document List Revision F dated December 09, 2017 or later accepted revision ICA-RH544-1009 INSTRUCTIONS FOR CONTINUED AIRWORTHINESS Revision 4 dated March 21, 2018.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>29/01/2018</td>
<td>Active</td>
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<tr>
<td>RAMI AEROSPACE INC.</td>
<td>STC</td>
<td></td>
<td>TCCA SH17-1</td>
<td>10064478</td>
<td>R44, R44 II</td>
<td>Replacement Seat Cushion/Backrest Installation &amp; Interior Accessories Installation of replacement seat cushion for pilot/co-pilot/pax, as well as interior accessories.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>MDL-RH544-1011 MASTER DOCUMENT LIST Revision D dated November 14, 2017 or later accepted revision ICA-RH544-1011 INSTRUCTIONS FOR CONTINUED AIRWORTHINESS Revision 3 dated November 13, 2017 or later accepted revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce noise/effect upon the airworthiness of the product.</td>
<td>29/01/2018</td>
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<td>1006486</td>
<td>RIAMM AEROSPACE INC.</td>
<td>TC88-7914-28</td>
<td>26/01/2018</td>
<td>STC88-7914-28</td>
<td>R44, R44 II</td>
<td>Installation of Replacement Headliner and Bulkhead Covering. Repair Bag.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>MDL-FRAM-2010 Master Document List Revision C, dated November 18, 2017, later accepted.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise approved design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/12/2018</td>
<td>Active</td>
<td>10064862</td>
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<tr>
<td>1006487</td>
<td>H4 AEROSPACE (UK) LTD</td>
<td>REV. 1</td>
<td>13/06/2019</td>
<td>CAA UK RESPONSIBILITIES</td>
<td>737-800, 737-900</td>
<td>Honeywell ADS-B Transponder Upgrade from TRA4B7 to TRA410AB Revision 1: including the interface with Rockwell Collins GLU905 and GLU925 MMRs as additional approved GPS position sources for ADS-B Compliance. In addition, the ICA were amended.</td>
<td>The Certification Basis for the original product as amended by the following or additional or alternative airworthiness requirements - Deviation(s): F-126: ADS-B Out. Extended Squitter Installation, issue 1. The requirements for environmental protection and the associated certified noise and/or emissions levels.</td>
<td>AFMS HAFA5M0047 rev 2 - ICA HAFC003 issue 3 - Modification Summary Sheet HA1875 issued for a later revision of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise approved design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/01/2018</td>
<td>Active</td>
<td>10064847</td>
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<td>1006494</td>
<td>KNSI AVIATION ENGINEERING LIMITED</td>
<td>REV. 1</td>
<td>26/01/2018</td>
<td>CAA UK RESPONSIBILITIES</td>
<td>A330-243</td>
<td>A330 – CABIN RECONFIGURATION.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List reference 10K030-MDL-001-16R (Revision 16).</td>
<td>None.</td>
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<td>10064944</td>
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<td>1006495</td>
<td>DASSAULT AIRCRAFT SERVICES (IAS)</td>
<td>FAA STO156NYN</td>
<td>D0036-150, Revision D, dated 22 March 2017 or later accepted.</td>
<td>FALCON 2000EX</td>
<td>FALCON 2000</td>
<td>Validation of FAA STC ST0038NYN Installation of a Data Link Communication System.</td>
<td>The Certification Basis for the original product as amended by the following or additional or alternative airworthiness requirements - Special Condition(s): CR1-F-2122-001: Flight Recorder and Data Link Recording. The following paragraph(s) at a later amendment: CS25 Amendment 18 of CS25.1459 and CS.25.1457.</td>
<td>FAA Flight Manual Supplement Document No FAA-M- D0036-150, Revision R, dated 03 June 2016. =&gt; Instructions for Continued Airworthiness Document No: F2XW- D0036-155R (Revision R), dated 09 May 2016. =&gt; later revisions.</td>
<td>This STC requires as a prerequisite, the installation of EASA STC 10048395 Rev. 1 and the installation of an approved Satellite Communication System that meets the requirements of AC 20-140B Prior to installation of this design change it must be determined that the interrelationship between this design change...</td>
<td>26/01/2018</td>
<td>Active</td>
<td>10064945</td>
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<td>1006496</td>
<td>LUFTNASSA TECHNIK AG</td>
<td>REV. 4</td>
<td>01/03/2024</td>
<td>CAA UK RESPONSIBILITIES</td>
<td>737-800, 737-700, 737-700, 737-800</td>
<td>Provisions for large KA band antenna and Link Communication System.</td>
<td>The Type and OSG Certification Basis for the original product as amended by the following or additional or alternative airworthiness requirements - Equivalent Safety Finding(s): B-101: Vibration / buffeting compliance criteria for large external antenna. The requirements for environmental protection and...</td>
<td>- Supplementary Type Definition Document ST0033-1920M issued 04 rev 00 - Airplane Flight Manual Supplement A-00-16/94-1AF revised 04 rev 03 - Instructions for continued airworthiness A-00-16/94 ICA issue revised 00 =&gt; later revisions of the above listed document(s) approved for the original product as amended by the EASA system.</td>
<td>Installation on aircraft models only that satisfy the configurations or criteria defined in document AU2013-21 and AU2023-51-E1-003. - This change cancels Alternative Method Of Compliance for EASA/Adopted FAA AD 2013-08-15, 2013-08-16 and 2013-19-23 for the areas affected by this modification.</td>
<td>01/03/2024</td>
<td>Active</td>
<td>10064496</td>
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<td>1006496</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STO4044CH</td>
<td>D0036-150, Revision D, dated 22 March 2017 or later accepted.</td>
<td>MYSTERE FALCON 50</td>
<td>MF50</td>
<td>Validation of FAA STC ST04044CH Installation of a FANS 1A CPDLC System</td>
<td>The Certification Basis for the original product as amended by the following or additional or alternative airworthiness requirements - Special Condition(s): CR1-F-5122-001: Flight Recorder and Data Link Recording. The following paragraph(s) at a later amendment: CS25 Amendment 18 of CS25.1437 and CS 25.1457 in the requirements for environmental protection and...</td>
<td>- Standard Boeing Master Data List No. 1029262, Revision D, dated 22 March 2016 =&gt; Airplane Flight Manual Supplement Document No 1029070, Revision 0, dated 22 March 2016 =&gt; =&gt; later revisions of the above listed document(s) approved for the original product as amended by the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise approved design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/01/2018</td>
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<td>10064496</td>
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<td>EASA STC No</td>
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<td>Rev. Level</td>
<td>Foreign/NAA</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10064501</td>
<td>Delta Engineering Corporation</td>
<td>REV. 1</td>
<td>FAA STC ST03420NY</td>
<td>EMB-155/165/165MP EMB-145/EMB-146/146ER EMB-135/135ER/EMB-135LR</td>
<td>Installation of a Business Class Commuter Interior.</td>
<td>Data Engineering Master Data List 10064504-2411 Revisions dated 15 June 2019 or later FAA approved revisions to Data Engineering Master Data List previously listed. The instructions for Continued Airworthiness (CR) and Electrical Wiring Interconnect System instructions for Continued Airworthiness and...</td>
<td>For Configuration 3 and 3 of the Master Data List EASA STC 10055137; 10055138. Prior to installation of this design change it must be determined that the interchangeability between this design change and any other previously installed design change and/or repair will introduce aero/roadside effect upon the aircraft.</td>
<td>11/07/2019</td>
<td>Active</td>
<td>10054070</td>
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<tr>
<td>10064504</td>
<td>Gama Aviation (Engineering) Limited</td>
<td>REV. 1</td>
<td>FAA STC 700620NY</td>
<td>GAMA AVIATION</td>
<td>Aspen EFD1000H: This change replaces the current analog Attitude Indicator (AI) and Directional Gyro (DG) and replaces them with the Aspen EFD1000H.</td>
<td></td>
<td></td>
<td>Signal coupling to autopilot (if installed) is not covered by this STC. Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce a noticeable effect upon the aircraft...</td>
<td>31/01/2018</td>
<td>Active</td>
<td>10064504</td>
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<tr>
<td>10064512</td>
<td>Lufthansa Technik AG</td>
<td>REV. 1</td>
<td>FAA STC 700620NY</td>
<td>EC135T1/EC135T2/EC135P1/EC135P2</td>
<td>Installation of the Dual GTN650/750 System into the EC135 helicopter.</td>
<td>Flight Manual Supplement EC135-07530-025-MFS, Revision 0 of later revisions of the above listed documents approved by EASA. Compliance Summary EC135-07530-027-RIV-00, R...</td>
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<td>10064516</td>
<td>Outerlink Corporation</td>
<td>REV. 1</td>
<td>FAA STC 700620NY</td>
<td>A91139 A81139</td>
<td>Installation of a Satcom and Flight Data Monitoring System KIT139-SAT-100.</td>
<td>- Supplemental Type Definition Document STD023-1702M Issue/Rev 1/0; - Airline Flight Manual Supplement A-00-1695-AM Issue/Rev 1/0; - Instructions for Continued Airworthiness Issue/Rev 1/0 of later revisions of the above listed documents approved by EASA.</td>
<td>- Only for activation of provisions certified by EASA STC 10064946. Installation on aircraft models only that satisfy the configuration/and criteria defined in Description Change AVE01-23 Rev. 5 or later approved revisions. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a noticeable effect upon the aircraft.</td>
<td>30/01/2018</td>
<td>Active</td>
<td>10064516</td>
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<tr>
<td>10064518</td>
<td>Outerlink Corporation</td>
<td>REV. 1</td>
<td>FAA STC 700620NY</td>
<td>A91139 A81139</td>
<td>Installation of a Satcom and Flight Data Monitoring System KIT139-SAT-100.</td>
<td>Outeralink Corporation Master Document List (MDL), Document number MDL-139-SAT-100, Revision - dated 8 November 2016 or later FAA approved revision. Operation must be in accordance with Outeralink Rotorcraft Flight Manual Supplemental (RFMS), Document number FMS 139A1001, Revision IN, FAA approved on 27 Apr...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce a noticeable effect upon the aircraft...</td>
<td>30/01/2018</td>
<td>Active</td>
<td>10064516</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10064523</td>
<td>GVR AEROSPACE (IRELAND) LIMITED</td>
<td>REV. 1</td>
<td>A320-214, A320-216</td>
<td>Airbus A320 - Installation of 150 Passenger Configuration.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): SC-D-5068 installation and smoke density requirements to seat material. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>FAA STC 10064523</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/01/2021</td>
<td>Active</td>
<td>10064523</td>
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<tr>
<td>10064542</td>
<td>BALLOONACY, LLC</td>
<td>FAA STC SB0447AT</td>
<td>SEE FAA ANL REV. A</td>
<td>Installation of the DB-III Omega</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Drawing List 5001, Revision B, dated 8 September 2017, or later FAA approved revision 2. FAA Approved Airplane Flight Manual Supplement, Drawing Number 5007 Revision IR, dated 8 September 2017, or later approved revision. 3. Instructions for Continued Airworthiness Document (CAD), Drawing Num...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and models...</td>
<td>02/02/2018</td>
<td>Active</td>
<td>10064542</td>
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<td>10064557</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SR09312RC-D</td>
<td>AS 350 B3 AS 350 B2</td>
<td>Installation of an Avylene 610 or 615 Traffic Advisory System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. Drawing List 350A5B-1760, Rev. H, dated 25 November 2015 for ODA approved revision 2. For Avylene 610- Rotorcraft Flight Manual Supplement 350FMS-177 dated 16 February 2009 or later ODA approved revision - Instructions for Continued Airworthiness 350MRS-020, revision dated 02 February 2009 or...</td>
<td>This STC is only applicable for post 12747 helicopters. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>02/02/2018</td>
<td>Active</td>
<td>10064557</td>
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<tr>
<td>10064558</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>REV. 1</td>
<td>AS 350 B3</td>
<td>Installation of a Garmin GTS 800 Traffic Advisory SystemRev. 1 adds the certification 35 ASB-1657-02.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: 27.810 (a)(7) (ii) Amdt 27-46; 27.773 Amdt 27-50; 27.1309 (a)(7) Amdt 27-46 and 27.1329 (f) Amdt 27-35; The requirements...</td>
<td>For the configuration 35GA2-1657-01- Aircraft must be AS 35B B post AMS07-4723 (TC G505H Installation). For the configuration 35GA2-1657-02- Aircraft must be AS 35B Post ECP OP-4862 - Aircraft must include previous installation approval of the GTX 345(R) Transporter. Prior to installation of...</td>
<td>25/01/2021</td>
<td>Active</td>
<td>10064558</td>
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<tr>
<td>10064559</td>
<td>AAR AIRCRAFT SERVICES</td>
<td>FAA STC ST04180CH</td>
<td>A330-243</td>
<td>Installation of Forward Left Hand, Right Hand, and Center Monuments.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>AMR Master Data List, 31725CR030, Revision B, dated 5 March 2017, or later FAA approved revision.</td>
<td>AAR STC 10064797 must be concurrently installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>27/02/2018</td>
<td>Active</td>
<td>10064559</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
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<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10064583</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SH70115SW-D</td>
<td>AS 350 B1/AS 350 B1Y/AS 350 B2/AS 350 B3 AS 350 BV/AS 350 D</td>
<td>Installation of a Nightvision or Starburst Searchlight.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Drawing List 355A82-174-09, Revision B. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/02/2018</td>
<td>Active</td>
<td>10064583</td>
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<td>10064584</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SR09511RC-D</td>
<td>AS 350 B2/AS 350 B3</td>
<td>Installation of a SMS Video Downlink System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Drawing List 355A43-1791, Rev. J, dated 24 November 2016 for later ODA approved revision 2. Rotorcraft Flight Manual Supplement 355FSM-525, Rev A, dated 16 June 2016 or later ODA approved revision 3. Instructions for Continued Airworthiness 355OMMS-031, Rev. 4, dated January 2016 or later FTW AES etc.</td>
<td>05/02/2018</td>
<td>Active</td>
<td>10064584</td>
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<td>10064585</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SR04909RC-D</td>
<td>MBB-BK117 C-2</td>
<td>Installation of IFR Single or Dual Pilot WAAS GPS Upgrade.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Drawing List 145A82-816B, Rev. M dated 12 September 2013 for later ODA approved revision 2. Rotorcraft Flight Manual Supplement 117FSM-237Rev. B, dated 20 September 2013 or later ODA approved revision 3. Instructions for Continued Airworthiness 145OMMS-020, Rev...</td>
<td>06/06/2018</td>
<td>Active</td>
<td>10064585</td>
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<td>10064586</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SR04908RC-D</td>
<td>EC 155 B1</td>
<td>Activation of Airbus Helicopters (formerly Eurocopter France) Provisional GPS with WAAS Vertical Guidance Rev. 1 of this STC is made to add the configuration 155A73-0021-02.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>2. Drawing List 155A82-816B, Rev. R dated 29 July 2016. For 155A73-0021-02 configuration (dual Garmin GPS system): Rotorcraft Flight Manual Supplement 155FSM-013, Rev. R, dated 19 Aug. 2013. Instructions for Continued Airworthiness 155OMMS-021, Rev. RI dated 16 May 2013. For 155A...</td>
<td>21/01/2021</td>
<td>Active</td>
<td>10064586</td>
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<tr>
<td>10064600</td>
<td>AIRBUS DEFENCE AND SPACE S.A.</td>
<td>REV. 5</td>
<td>A330-243</td>
<td>A330 MRRT ROKAF STC Conversion of A330-200 Series A/C to A330 Multi Role Transport Tanker (MRTT) ROKAF (Republic of Korea Air Force/Rev. 1 adds the STOPS capability of the basic aircraft to MRTT ROKAF/Civcraft/Rev. 2 introduces ALS Part 2 Supplements for MRTT STCS based on A330 ALS Issue 19 After the rel...</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following additional or alternative information: - For 155A73-0021-01 configuration (dual Garmin GPS with 400W system)(&gt;-&gt;) &gt; this STC is limited to helicopters equipped with the following (a) Type Certified Dual Pilot/Single Pilot or Single Pilot Dual Garren GNS 430/Installation and Type Certified MEGHAS part number C10872X3S-12 or C10872X2S-350-46 displays must be installed. This installation is not compatible with Helicopter T...</td>
<td>18/04/2023</td>
<td>Active</td>
<td>10064600</td>
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<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>STC Holder</td>
<td>Limitations/Conditions</td>
<td>Active</td>
<td>V35B (TC), G36, A36TC, SCANDINAVIAN IKHANA GROUP, 10064631</td>
<td>Models</td>
<td>328 SUPPORT SERVICES GmbH</td>
<td>ADS-B OULT - Rockwell Collins Pro Line 21 Avionics on GULFSTREAM 150</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements.- Deviation(s): CRI F-01. Continuity Requirements for ADS-B function. - The following paragraph(s) at a later amendment: CS-ACN, Initial Issue: CS-ACNs D ADSB. 001, CS ACNs D...</td>
<td>Master Document List (MDL), Document No. MOL138201, Rev. 0, BAFM Supplement, Document No. AFMS1683420001, Rev. B, MMSL, Supplement, Document No. MMSL1683430001, Rev. B</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/02/2018</td>
<td>Active</td>
<td>10064629</td>
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<td>10064606</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>206L-1</td>
<td>Garmin single GTN 650 or dual GTN650/750 installation.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Subpart B Communication (COM) Section 1 of CS-ACN. Amendment dated 08.05.2013.</td>
<td>Operation in accordance with Scandinavian Avionics design AFMSupplement 2107136-AM. Initial Issue 6 dated 17.02.2018; 2107136-AM. Single Garmin GTN 650, Initial Issue 2 dated 07. Feb. 2018; or later revisions of the above listed documents approved by EASA.</td>
<td>Installation of this STC requires prior embodiment of the IFR configuration 207-705-001. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/02/2017</td>
<td>Active</td>
<td>10064606</td>
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<td>10064614</td>
<td>AIRBUS HELICOPTERS, Inc</td>
<td>FAA STC SR0496AR-C</td>
<td>EC135 T2/ EC135 T2+</td>
<td>Installation of IFR Single or Dual Pilot IAAS GPS upgrade.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. Drawing List 135MS-021P, Rev. P dated 31 March 2016 or later ODA approved revision 2. Rotorcraft Flight Manual Supplement (RFMS) 135MS-034 Rev. E dated 19May 2016, for Garmin GNS-430/AW installations, or RFMS 135FS-035 Rev C dated 18May 2016, for Garmin GNS-530/AW installations or later ODAs. Prior to or concurrently with the installation of this STC, the FAA approved Single Pilot/Single or Single Pilot Dual Garmin GNS 430W Installation, or FAA approved GNS 530W substitution(s) for the Type Certified GNS 430W(s), or Rotorcraft Flight Manual Supplement, Document No. E-2931, Rev. 09E dated 21 June 2017.</td>
<td>06/02/2018</td>
<td>Active</td>
<td>10064614</td>
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<td>10064624</td>
<td>AIRBUS S.A.S.</td>
<td>A320-216</td>
<td>MOD A0009 - IBE a320 - ISPS</td>
<td>Activation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Modification Approval Sheet, ref. doc. MAS 1 A0009, Issue 1 dated 04/01/2008.</td>
<td>None</td>
<td>13/02/2018</td>
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<td>10064624</td>
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<td>10064630</td>
<td>IKHANA GROUP, LLC</td>
<td>REV. 3</td>
<td>FAA STC SA0942ALC</td>
<td>DHC-6 SERIES 200 DHC-6 SERIES 100</td>
<td>Installation of Pratt &amp; Whitney Canada PT6A-27 engines rated at 820 SHP/412.3kW at 2,000 ft (610 m) for Single or Single Pilot with Scandinavian Avionics Avionics Inc. Design AFMSupplement: - Deviation(s): CRI A36TC F-01. Continuity Requirements for ADS-B function. - The following paragraph(s) at a later amendment: CS-ACN, Initial Issue: CS-ACNs D ADSB. 001, CS ACNs D...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. IKHANA Aircraft Services Drawing List ER218-2027, Revision B[1] dated 30 May 2017, or later FAA approved revision.2. IKHANA Aircraft Services Master Data List ER218-2016, Revision NC[2] dated 21 June 2017, or later FAA approved revision.</td>
<td>Installation of this STC does not allow increase in Max Gross Weight. This modification has been substantially to a non-acoustic change part 14 CFR part 21.53 (b) Prior to installation of this design change it must be determined that the interrelationship between this design change and any other approval(s).</td>
<td>08/09/2021</td>
<td>Active</td>
<td>10064630</td>
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<tr>
<td>10064631</td>
<td>MT-PROPELLER ENTWICKLUNG GMBH</td>
<td>REV. 1</td>
<td>V38B (TC), G38, A38TC, B36TC</td>
<td>Installation of 3-blade constant speed propeller MTV-B-D280-56 on Textron Aviation Inc. Bonanza Models</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-36. Amendment 36-30 of this certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise...</td>
<td>According to MT-Propeller Master Documentation List No. E-2934, Rev 0 to be operated in accordance with any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>According limitation section of AFM/SP for installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2018</td>
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<td>EASA STC no.</td>
<td>B/E AEROSPACE, Inc.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>10064641</td>
<td>FAA STC358AT</td>
<td>737-200/737-300/737-300F</td>
<td>Installation of the Rockwell Collins Automatic Dependent Surveillance – Broadcast (ADS-B) Out System</td>
<td>- Master Drawing List (MDL): 2-171676-001, Rev. 0 dated 25 September 2011</td>
<td>FAA approved revision. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/07/2021</td>
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<td>10064656</td>
<td>GAMA AVIATION (ENGINEERING) LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B300, B300C, B200 B200C, B200C, B200CGT, B200CT</td>
<td>Gama Modification GAEL-1548 - Installation of two Garmin GPS Antennas GA-36 and GA-37.</td>
<td>- To be installed according to GAMA Document GAEL-1548MDL-01(15) (&gt;issue A, dated 26 January 2018). To be maintained according to Instructions for Continued Airworthiness (ICA), GAAM Document GAEL-1548ICA-01(15) (&gt;issue 1, dated 26 January 2018) for later revisions of the above listed documents approves...</td>
<td>14/02/2018</td>
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<td>10064675</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SR0915RC</td>
<td>505</td>
<td>Installation of Cargo Mirror Kit.</td>
<td>The Certification Basis for the original product remains applicable to the ICA- The scope of this STC modification does not introduce any adverse effect upon the airworthiness of the product.</td>
<td>15/02/2018</td>
<td>Active</td>
<td>10064675</td>
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<td>10064677</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SR0914RC</td>
<td>407</td>
<td>Installation of a 407 Imaging System Kit.</td>
<td>The Certification Basis for the original product remains applicable to this certificate and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/02/2018</td>
<td>Active</td>
<td>10064677</td>
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<td>10064687</td>
<td>FLYHT AEROSPACE SOLUTIONS Ltd.</td>
<td>TCCA SA13-02 REV. 03</td>
<td>737-700, 737-800 737-300, 737-400, 737-500</td>
<td>AFRS 228 Automated Flight Information Reporting System Installation, in accordance with FLYHT MDL 121-0075 Rev. F (for 737-300, 737-400, 737-500) and FLYHT MDL 121-0065 Rev. AC (for 737-700, 737-800).</td>
<td>- The airplane modified by this STC must be operated in accordance with the AFMS - maintained in accordance with the ICA. The scope of this STC modification does not introduce any adverse effect upon the airworthiness of the product...</td>
<td>15/02/2018</td>
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<td>Rev. level</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10064670</td>
<td>BELL</td>
<td>Active</td>
<td>FAA STC 407-998-001</td>
<td>407</td>
<td>External Hoist Kits and Hoist Retrofit Kit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Aircraft Accessories, Inc. Drawing List, Report No. AA-97005; Rev. 11, dated 17 May 2008. For External Hoist Kit Installation Instructions Report No AA-97006, Rev. K, dated 22 March 2004. For Hoist Retrofit Kit 407-998-001, Installation Instructions Report AA-00565, No Revision, dated 16 May 4...</td>
<td>15/02/2018</td>
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<td>10064841</td>
<td>MORSOON</td>
<td>Limited</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>HS-125 SERIES 700E</td>
<td>Introduction of ADS-B Out functionality.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: FAR 25.1529 Amdt 54-FAR 25.1581 (a) &amp; (b) Amdt 72-CS-ACNS Initial Issue (Subpart D Section 4). The requirements for...</td>
<td>Installation, Operation and Maintenance according to MorsonProjects Limited. Change Report MP0010A issue 1, dated January 2018.&lt;(&gt;)(&lt;)or later revisions of the above listed documents approved by EASA.</td>
<td>16/02/2018</td>
<td>Active</td>
<td>10064841</td>
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<td>10064885</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>REV. 3</td>
<td>EC 175 B</td>
<td>EC175 EMS equipment installation. The change consists of different layouts (one or two patient configurations) and can include the following components: Medical equipment carrier (MEC), Oxygen station, Stretcher (including footrest), Locker system. Reason for Revision 1: the following manuals/documents h...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List, Doc. No.: MDL-EC175-25-20-104-000-601 to 638 REV BF. Flight Manual Supplement, Doc. No.: FMS-EC175-EMS-000-006, Rev. 2. Instructions for Continued Airworthiness, Doc. No.: ICA-EC175-EMS-000-Rev. Installation Manual, Doc. No.: IM-EC175-EMS-000Weight and Balance Manual, Doc. No.: WBA-EC175-E...</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/12/2019</td>
<td>Active</td>
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<td>10064998</td>
<td>365 AEROSPACE DQS LIMITED</td>
<td>REV. 3</td>
<td>A320-214</td>
<td>EC175 EMS equipment installation. The change consists of different layouts (one or two patient configurations) and can include the following components: Medical equipment carrier (MEC), Oxygen station, Stretcher (including footrest), Locker system. Reason for Revision 1: the following manuals/documents h...</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval...</td>
<td>Defined within approved Model List (AML) document, Ref. 3ML-2016 Issue C, dated 25 June 2024 or later revisions of the above listed documents approved/accepted under the EASA system.</td>
<td>The system is not approved for CAT IIIb approach flight phases. Prior to installation of this certification/ approval, the installation...</td>
<td>28/06/2024</td>
<td>Active</td>
<td>10064998</td>
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<td>10064703</td>
<td>STANDARDAERO ATLANTIC USA, INC.</td>
<td></td>
<td>FAA STC 407-998-001</td>
<td>AS 350 B4</td>
<td>Installation of a Crash Resistant Fuel System (CRFS).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments/ revisions on FAA STC SR04202A. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product...</td>
<td>1. Master Document List (MDL) AS350-1420-MDL Revision A, dated 31 October 2017, or later FAA approved revision 2. Vector Aerospace “Instructions for Continued Airworthiness, for the Crash Resistant Fuel System (CRFS)”, Document Number 407-998-001-500-ICA, Revision D, dated 3 November 2017, or later FAA...</td>
<td>This installation has been shown compliant with the requirements for ACROFS when concurrently installed with the rotorcraft manufacturer’s/aro...</td>
<td>20/06/2018</td>
<td>Active</td>
<td>10064703</td>
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<tr>
<td>10064716</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td></td>
<td>USB Power on B767</td>
<td>767-400ER</td>
<td>USB Power on B767</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: RFL CCL-NAV0573 Rev 2. The requirements for environmental protection and the associated certified noise and/or emissions lev...</td>
<td>AFM – doc. API-NAV0573 Rev 1 – doc. ICA-NAV0573 Rev 0 or later revisions of the above listed documents approved by EASA Certification Program docs. P-2 NAV0573 Rev. 20CL – doc. CCL-NAV0573 Rev 2</td>
<td>None</td>
<td>20/02/2018</td>
<td>Active</td>
<td>10064716</td>
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</table>
10084718 STC Holder: AIRBORNE TECHNOLOGIES GmbH
Rev.: REV. 2
Foreign/NAA STC ref.: A97105SP
Model: AW109-Mission Configuration including Trakia A800 searchlight, camera/flight system with MO-15 HSE and a cabin workstations (STC-AT-102)

Description: Certification basis for the original product as amendability of flight testing additional or alternative airworthiness requirements - Special Condition (s): CRI F-01 Special Condition for Lithium-Ion Battery Installation - The following paragraphs(s) at a later amendment: CS27.49 Amdt 1. The requirements for...

Certification basis: Rotorcraft Flight Manual Supplement STC-102-0- AFM rev 05, dated 18 October 2018, Certification Maintenance REQUIREMENTS as per paragraph 4.3 of STC-102-0-AFM at rev 03 dated 06 January 2018 or later revisions of the above listed document(s) approved acceptably the EASA system Certification Program.

Limitations/Conditions: Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effects upon the airworthiness of the product.

Date of Issue: 02/01/2021
Status: Active
System Record: 10084718

10084719 STC Holder: BOMBARDIER Inc.
Rev.: REV 1
Foreign/NAA STC ref.: CL-600-2B16 (604 VARIANT)
Model: EASA Validation of Transport Canada STC SAT-17-19 at Issue 1

Description: Certification basis for the original product as amendability of flight testing additional or alternative airworthiness requirements - Special Condition (s): BOMBARDIER Inc.

Certification basis: Modification to be completed in accordance with STC Certification basis for installation of Securaplane Technologies High Definition External Camera System (CMX-HD) to replace existing standard High Definition External CameraSystem Installation of Securaplane Technologies SATCOM Datalink Interface

Limitations/Conditions: The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.

Date of Issue: 02/01/2018
Status: Active
System Record: 10084719

10084720 STC Holder: SECURAPLANE TECHNOLOGIES, Inc.
Rev.: REV. 1
Foreign/NAA STC ref.: GV-SP
Model: Installation of Securaplane Technologies High Definition External Camera System (CMX-HD) to replace existing standard definition camera system

Description: Certification basis for the original product as amendability of flight testing additional or alternative airworthiness requirements - Special Condition (s): FAA STC ST01420C


Limitations/Conditions: EASA STC 10015697 (FAA STC ST02989DRT-D) for GV-SX Series aircraft must be installed prior to or in conjunction with this STC.

Date of Issue: 02/01/2018
Status: Active
System Record: 10084720

10084726 STC Holder: B/E AEROSPACE LIMITED
Rev.: REV 1
Foreign/NAA STC ref.: BD-700-1A10
Model: EASA validation of Transport Canada STC C-LSA14-035D at issue 2Installation, Complete Custom Aircraft Interior

Description: Certification basis for the original product as amendability of flight testing additional or alternative airworthiness requirements - Special Condition (s): EASA CRI D-300 at issue 3 - Deviation – Environmental Protection and the applicable to this certificate/ approval.

Certification basis: Modification to be completed in accordance with Modification DataSummary J122000.

Limitations/Conditions: The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.

Date of Issue: 02/01/2018
Status: Active
System Record: 10084726

10084730 STC Holder: B/E AEROSPACE LIMITED
Rev.: REV 1
Foreign/NAA STC ref.: BD-700-1A10
Model: EASA validation of Transport Canada STC C-LSA14-035D at issue 2Installation, Complete Custom Aircraft Interior

Description: Certification basis for the original product as amendability of flight testing additional or alternative airworthiness requirements - Special Condition (s): EASA CRI D-300 at issue 3 - Deviation – Environmental Protection and the applicable to this certificate/ approval.

Certification basis: Modification to be completed in accordance with Modification DataSummary J122000.

Limitations/Conditions: The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.

Date of Issue: 02/01/2018
Status: Active
System Record: 10084730

List of EASA Supplemental Type Certificates
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<th>STC Holder</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>STC Holder</td>
<td>550, 560, S550, 500</td>
<td>VHF/GPS Antenna Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment/CAO-AM of Airworthiness Certificate or Approval for Airworthiness. Requirements as listed in EASA STC, Document ref: 211373 CO, Amendment 10064734, Issue 1. The requirements for envir...</td>
<td>- Engineering Order &amp; Certification Program, Doc Ref: 211373-EO, Issue 1, Compliance Record, Doc Ref: 211373 COR, Issue 1, for later revisions of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/02/2018</td>
<td>Active</td>
<td>10064732</td>
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<tr>
<td>FLYHT AEROSPACE SOLUTIONS Ltd.</td>
<td>TCCA SA12-67 ISSUE 3</td>
<td>AFIRS 228 Automated Flight Information Reporting System Installation in accordance with FLYHT Master Document List 121-0068 Rev. M dated 21/02/2018</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment/CAO-AM of Airworthiness Certificate or Approval for Airworthiness. Requirements as listed in EASA STC, Document ref: 211373 CO, Amendment 10064734, Issue 1. The requirements for envir...</td>
<td>- FLYHT ICA Document ref. 153-0059 Rev. E and FLYHT ICA Document Ref. 157-0015 Rev. B, dated February 2017, FLYHT ICA Document Ref. 153-0059 Rev. E and FLYHT ICA Document Ref. 157-0015 Rev. B, dated 26 March 2015, and FLYHT ICA Document Ref. 157-0015 Rev. B, dated 5 January 2018, for later revisions of the above listed documents approved by EASA...</td>
<td>Condition(s) - The aeroplanes modified by this STC must be - operated in accordance with FLYHT AFM 5 Rev. 0 and later approved revisions, maintained in accordance with FLYHT ICA 153-0059 Rev. E and FLYHT ICA Document Ref. 157-0015 Rev. B, or later approved revisions. Notes - This STC does not apply.</td>
<td>21/02/2018</td>
<td>Active</td>
<td>10064734</td>
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<tr>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>REV. 1</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of Monogram Systems’ In-Flight Traffic Compactor Revision 1: TrafficAware Kit Installation Instructions, Boeing 767-200/300; Ref. 155-0011 Rev. B, dated 10/03/2013, and any other previously installed design change and/or repair. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>1. Monogram Systems’ Master Data List TR48800, Revision C, dated 9 September 2011.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2021</td>
<td>Active</td>
<td>10064735</td>
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<tr>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>737-800</td>
<td>Tailored SFL TV Antenna Installation/RO Design FAA STC ST01850NY (not EASA validated), installed on upper fuselage between STA 600C and STA 540, certifying against EASA Approval Basis</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Findings (e.g.)- 90-2001 Vibrations / Buffeting Criteria for large external antennas/ installations. The requirements for environmental protection and the associated...</td>
<td>- Master Data List MDL-GE01-5300 Revision A, Dated 20.02.2018, Military Flight Manual Supplement Amendment 1 to Airworthiness Certification Program, FAA-GE01-5300, and other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Per Airworthiness Limitations Section of Instructions for Continued Airworthiness for VHF GPS Antenna Installation, Doc Ref: 211373 COR, Issue 1, for later revisions of the above listed documents approved by EASA.</td>
<td>23/02/2018</td>
<td>Active</td>
<td>10064772</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>REV. 1</td>
<td>P.68R “VICTOR” P.68R “VICTOR”</td>
<td>TCAS VII design change approved the upgrade of a Garmin GTS Processor, TrafficAdvisory (TAS) GTS 875G to a Traffic Collision Avoidance System (TCAS III) (GTS 8000) together with the installation of GTX300Transponder and GSB antenna to facilitate TCAS II capability. Changes of this STC STC-AT-104A 1...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 23 Amdt 3, CS 23.301; CS 23.303; CS 23.307; CS 23.337; CS 23.625(a); CS 23.771(a); CS 23.876(a); CS 23.1301; CS 23.1305 2...</td>
<td>The aircraft must have been fitted with the TC holder’s Garmin GTS8000 prior to the upgrade of the system under this approval. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/03/2023</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>IVX AEROSPACE, Inc. - FSI</td>
<td>REV. 5</td>
<td>FAA STO251SE</td>
<td>777-200</td>
<td>British Airways Boeing model 777-200 series interior reconfiguration Revision 5 updates the Limitations/Conditions section.</td>
<td>Master Drawing ref. BIFS657-D01 rev. AM dated 19/09/2020/Supplement to Instructions for Continued Airworthiness ref. BIFS657-D15 rev. G dated 23/07/2020/AIRMS ref. TPI18-865/AFM initial revision dated 03/09/2020/Other later revisions of the above listed document(s) approved/ accepted/modified of EASA in a...</td>
<td>&quot;Prior or concurrent installation of the following modification(s) anequired with this STC (aircraft applicability is identified within the following documentation)&quot; - British Airways Service Bulletin BA-777-25-0078, Rev. 00, dated September 12, 2018 and Concession Engineering Design Instruction/Approv...</td>
<td>28/09/2020</td>
<td>Active</td>
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<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td></td>
<td>FAA STC ST01794W</td>
<td>SEE FAA AML, REV. A System Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product aneunchanged and remain applicable to this certificate/ approval.</td>
<td>1. Airplane Flight Manual Supplement as specified by FAA Approved Model List (AML) RAC-051. Revision (-)2. Instructions for Continued Airworthiness as specified by FAA Approved Model List (AML) RAC-051. Revision (-) or later FAA approved revisions.</td>
<td>Prior EASA approved installation of a Rockwell Collins TCAS II system is a prerequisite to this STC. Model specific limitations are documented within AML RAC-051. Revision (-) or later approved revisions. Prior to installation of this design change it must be determined that the interrelationship between...</td>
<td>02/03/2018</td>
<td>Active</td>
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<tr>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>REV. 8</td>
<td>FAA STC SA02571SE</td>
<td>Installation of Garmin G500/G600 TXi Integrated Avionics System Primary Flight Display, Multi-Function Display, and Engine Indicating System/Diplay. Revision 1 amends the Master Drawing List, Equipment List, AFMS, ICA and Install Manual. Revision 2 is a correction of a typographical error. Revision 3 a...</td>
<td>For Aircraft in FAA AML to FAA STC SA02571SE. Garmin Master Drawing List (MDL) 055-00795-C2 Revision 1, dated 17 November 2012. Garmin Airplane Flight Manual Supplement (AFMS) 190-01717-B2 Revision1, dated 17 November 2017. Additional Garmin AFMS for the Socata TB1 700/850 (to address RVSM aspects).</td>
<td>AAR Master Data List 122, dated 26 June 2014/or later FAA approved revision.</td>
<td>This EASA STC is valid for aircraft types and models approved by EASA and listed on the FAA AML to FAA STC SA02571SE dated 15 March 2023 and for aircraft types and models listed in FAA AML to FAA STC 10064794/Attatched to this approval (which are only approved in EASA System &quot;EASAonly models&quot;) Prior to installation of this design change it must be determined that the interrelationship between...</td>
<td>21/07/2023</td>
<td>Active</td>
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<tr>
<td>AAR AIRCRAFT SERVICES</td>
<td></td>
<td>FAA STC ST0402CH</td>
<td>Installation of 278 (19BC/68PY/192EY) Passenger Seating Arrangement.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product aneunchanged and remain applicable to this certificate/ approval.</td>
<td>AAR Master Data List 177, dated 24 May 2016 or later FAA approved revision.</td>
<td>Lufthansa Technik AG STC 10024373 must be concurrently installed. Prior to installation of this design change it must be documented accurately to determine its interrelationship between this design change and any other previously installed design change/ and/ or repair or maintenance, etc.</td>
<td>27/02/2018</td>
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<tr>
<td>P.P.G. INDUSTRIES, Inc.</td>
<td></td>
<td>PQ1015CE SUPPLEMENT NO. 122</td>
<td>CL-600-2024 (RJ SERIES 900) CL-600-2120 (RJ 700-701-702) CL-600-2819 (RJ SERIES 100) CL-600 Main and Side Windsheid PPG Windshield/PM - CL-600</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product aneunchanged and remain applicable to this certificate/ approval.</td>
<td>- FAA - Parts Manufacturer Approval, letter PNA No. PQ1015CE, dated 22/08/2014/FAA PQ1015CE, Supplement 122, dated 26 June 2014/ Other later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/0417 (or subsequent revisions off this decision) and/or FAA...</td>
<td>None</td>
<td>27/02/2018</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10064803</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>737-700</td>
<td></td>
<td>Certification for Commercial Operation / The modification consists of removing the existing limitations for private use to allow commercial operation of the aircraft. For this purpose, physical modifications of the aircraft have been performed and substantiated through AMAC Minor modification AY07-2520...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment: Appendix III of CS-25 Amendment 19: S25.f(1); S25.10; S25.20; S25.30; S25.40(a); S25.50. The requirements for environment...</td>
<td>AFMS-AV07-2520, Airplane Flight Manual Supplement, Rev. A</td>
<td>airworthiness and/or repair will be accomplished prior to the installation of this modification. The requirements for environmental protection and the associated certified noise...</td>
<td>27/02/2018</td>
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<tr>
<td>10064804</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>REV. 1</td>
<td>HAWKER 800XP</td>
<td>ADS-B Out Upgrade</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-ACNS Initial Issue/Subpart D Section 4:The requirements for environmental protection and the associated certified noise...</td>
<td>Installation, Operation and Maintenance in accordance with Compliance Report CR-RH10-GOM23-001, issue R1 dated 3rd June 2020 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Limited to: BAe 125 Series 1000A and 1000B Serial Numbers 25681, 258159, 259044 through 259047, 294902 to 294904, 295850. When equipped with Honeywell STG-800 avionics given the following configuration: Honeywell Dual Integrated Communication Unit - RCU-800, PN:751070...</td>
<td>01/06/2020</td>
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<tr>
<td>10064806</td>
<td>ELECTRONICS INTERNATIONAL, Inc.</td>
<td>FAA STC SA02155E</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of Electronics International Inc. Glass Panel Engine Monitor (with In-Flight Data) MVP-50T</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Installation Instructions, Document No. II 12110161; Revision D, dated 15 June 2011 or later FAA approved revision 2. Airplane Flight Manual Supplement (AFMS) Document Number AFM0271-101, Revision IR, dated 27 June 2011, or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and models...</td>
<td>01/03/2018</td>
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<td>10064808</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC</td>
<td>REV. 1</td>
<td>A321-231</td>
<td>Cabin Densification of British Airways A321-231 Aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Summary, Northwest Aerospace Technologies, Inc. document No.6004-090, Revision 1, dated 3 January 2018; for later revisions of the above listed document(s)...</td>
<td>The following Minor Change Approved Service Bulletins shall be concurrently installed: British Airways SB A321-25-005; Airbus SB42320-63-1348, SB25-1 (SGV); AIB Aviation SBs 321 BM2AGAVO-25-1701, A23-1-25-016; ZSFR SB 160-25-174; HAA SB44142405. Prior to installation of this design change...</td>
<td>06/04/2018</td>
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<td>10064809</td>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FAA STC ST01005NY</td>
<td>FALCON 200 EX FALCON 2000</td>
<td>Validation of FAA STC ST01005NY installation of Meggitt Avionics Integrated Secondary Flight Display (GFD).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List No. F3MV-D0057-055, Revision B, dated 31 October 2016 Airplane Flight Manual Supplement Document No.: F3MV-D0057-150, Revision R, dated 02 June 2016 Instructions for Continued Airworthiness Document No.: F3MV-D0057-155, Revision A, dated 21 Oct 2016 or later revisions of the above listed document(s)...</td>
<td>This STC requires as a prerequisite, the installation of Dassault Aviation Modification M3152: Meggitt Digital Standby unit (Meggitt M4C). Prior to installation of this design change must be determined that the interrelationship between this design change and any other previously installed design change...</td>
<td>28/02/2018</td>
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<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>MYSTÈRE FALCON 50</td>
<td>Validation of FAA STC ST03587NY Installation of Meggitt Avionics Integrated Secondary Flight Display (SFD).</td>
<td>Master Data List No.</td>
<td>F50/WX-DO43-005, Revision A, dated 31 October 2016.</td>
<td>20/02/2018</td>
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<td>DASSAULT FALCON JET CORPORATION</td>
<td>MYSTÈRE FALCON 900</td>
<td>Validation of FAA STC SA70445W-D Antenna Installation.</td>
<td>Drawing List Report No.</td>
<td>F90-01055, Revision AD, dated 13 July 1995 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/or the Technical Implementation Procedures of EU/EASA Bilateral Agreement...</td>
<td>01/03/2018</td>
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<td>BABCOCK MISSION CRITICAL SERVICESDESIGN AND COMPLETIONS LIMITED</td>
<td>CVR/FDR Installation/NVIS on EC135</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>Master Data List No.</td>
<td>EASA-AECJ-2005-051, Issue 1 dated 07 October 2015.</td>
<td>01/02/2018</td>
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<td>AEROTECH &amp; CONCEPT</td>
<td>A330-342</td>
<td>Aircraft Flight Manual Supplement S.OAEE.01/1875-AM2 Revision 0 dated 01/03/2018.</td>
<td>-PMF Approval Supplement AE17161FMS-Per Instruction for Continued Airworthiness AE17161CA-Pre-requisite: Aerotec Change AE16077 MAM Prior to installation of this design change it must be determined that there are no adverse effects upon the airworthiness of the aircraft product.</td>
<td>01/03/2018</td>
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<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>IFE UpgradeRev 1: Transfer of STC from SR Technics Switzerland to grnWING Engineering Partners GmbH</td>
<td>Aircraft Flight Manual Supplement S.OAEE.01/1875-AM2 AI.</td>
<td>Prior to installation of this design change it must be determined that there are no adverse effects upon the airworthiness of the aircraft product.</td>
<td>21/04/2021</td>
<td>Active</td>
<td>10064621</td>
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**List of EASA Supplemental Type Certificates**

An agency of the European Union
List of EASA Supplemental Type Certificates

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<tr>
<td>FLYHT AEROSPACE SOLUTIONS Ltd.</td>
<td>10064848</td>
<td>-</td>
<td>TCMA STO NO. SA14-39 ISSUE 2</td>
<td>757-200</td>
<td>AFIRS 228 Automated Flight Information Reporting System Installation in accordance with FLYHT Master Document List 121-0076 Revision G</td>
<td>The Certification Basis for the original product as amended by following additional or alternative airworthiness requirements: Aircraft TCDS with the addition of FAR 29.869(a) at Amendment 25-72, FAR25.981 at Amendment 25-102 and 14 CFR 26 at Amendment 26-674a requirements for environmental products.</td>
<td>FAA Approved Airframe document AFIRS 228 Installation, Boeing 757-200, Ref. 10064848, dated 05 March 2018; or later revisions of the same.</td>
<td>05/03/2018</td>
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<td>10064869</td>
<td>ANJOU AERONAUTIQUE S.R.L.</td>
<td>-</td>
<td>-</td>
<td>737-800</td>
<td>Replacement of seatbelts on 16G aircraft seat</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>05-900428-MDL-001, Rev. 05 dated 05 March 2018 “Master Data List” or later revisions of the above listed documents approved by EASA.</td>
<td>06/03/2018</td>
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<td>10064887</td>
<td>AVATION FABRICATORS, Inc.</td>
<td>FAA STC SA128RM</td>
<td>501/551</td>
<td>Installation of stretchers.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Branson Engineering Company, Inc. Drawing List No. 90132, Revision C dated 13 March 1978 or subsequent FAA approved revision or according to Aviation Fabricators STC Drawing List AF-215, Revision B, dated 25 March 2003 or later FAA approved revision 2. Airplane Flight Manual Supplement dated 19 Dec.</td>
<td>Interchange provisions apply only to model 551. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/03/2018</td>
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<tr>
<td>10064889</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 1</td>
<td>B100</td>
<td>MTV-27(-j) on Beech B100</td>
<td>The Certification Basis for the original product as amended by following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment CS-36, Amdt 4, FAR 36Amdt 36-30 This certificate/approval involves a change to the requirements for environmental products.</td>
<td>According to MT-Propeller Master Documentation List No. E-3224, Rev 3 To be operated in accordance with MT-Propeller AFM No. 3224, Rev 2 To be installed i.a.w. Installation Instructions Doc. No. E-3227, Rev 2 To be maintained i.a.w. Instruction for continued airworthiness Doc. No. E-3228, Rev...</td>
<td>According to limitation section of AFM-S.Prior to installation of this design change/repair it must be determined that the interrelationship between this design change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/11/2019</td>
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<td>10064891</td>
<td>AIRFILM CAMERA SYSTEMS, INC.</td>
<td>FAA STC SR02660LA</td>
<td>SEE FAA AML</td>
<td>Installation of Nose Utility Mount.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA approved Airframe Camera Systems documentation titled Camera Systems Installation Instructions lists FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/03/2018</td>
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List of EASA Supplemental Type Certificates

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<td>CERT WORKS, LLC</td>
<td>REV. 2</td>
<td>FAA STC SR00826DE</td>
<td>SEE FAA-APPROVED MODEL LIST</td>
<td>Installation of Honeywell Sky Connect Tracker III system Rev. 1. Revised to add Leonardo S.P.A. AW169 Rev. 2. Revised to add Airbus Helicopters SA365N, SA365N1, AS365N2, AS365N3, EC155B1, EC155B1. The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if included on FAA STC SR00826DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. For Configuration 1: The change is defined through AMAC Master Data List MDL-A3201-1-5300, Revision C. The aircraft shall be modified in accordance with AMAC Engineering Order EO-A3201-1-5300, Revision B and associated documents. The aircraft shall be operated in accordance with AMAC Airplane Flight...</td>
<td>Document FAA-AMC001-1006, Master Data List (MDL); Part 29 AML SkyConnect Tracker III Installation, Revision IR dated 1 December 2018 or later FAA approved revisions. FAA approved Rotorcraft Flight Manual Supplement (RFMS). Refer to AML-SR00826DE for model specific RFMS document instructions for Continued Improvement. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/02/2020</td>
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<td>10064902</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>REV. 7</td>
<td>A320-231/-232 A321-211/-212/-213/-214/-215/-216 A320-231/-232/-233/-235 A320-231/-232/-233/-235 A320-231/-232/-233/-235 A320-231/-232/-233 A321-211/-212-213/-214/-215/-216 A319-115/-116/-117/-118/-119/-120/-121/-122/-123 A318-115/-116/-117/-118/-119/-120/-121/-122/-123</td>
<td>Installation of the Honeywell JetWave Ka-Band Antenna Revision 1: Extension of EASA STC 10064988 to include the A320-211, A320-212, A320-213, A320-214, A320-215, A320-216, A320-217, A320-218, A320.223, A320-225, AirbusModels Revision 2: Extension of EASA STC 10064988. Rev. 1 to include the A320-215N-252SN, A320-252SN. The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - Equivalent Safety Finding(s): CRI B-01 Vibration/buffeting compliance criteria for large external antenna installation. - The following paragraph(s) at a late...</td>
<td>For Configuration 1: The change is defined through AMAC Master Data List MDL-A3201-1-5300, Revision C. The aircraft shall be modified in accordance with AMAC EngineeringOrder EO-A3201-1-5300, Revision B and associated documents. The aircraft shall be operated in accordance with AMAC Airplane Flight...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/06/2024</td>
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<td>E.A.D AEROSPCE AIRWORTHINESS</td>
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<td>777-300, 777-300ER 777-200, 777-200R</td>
<td>Transmitting Portable Electronic Devices (T-PED) tolerance demonstration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Master Drawing List, EAD Aerospace document no. 01554-EAS-A-MDL-Revision C. dated 12 March 2018. Instructions for Continued Airworthiness. EAD Aerospace document no.01554-EAS-A-ICA-01, Revision B, dated 27 February 2018. Aircraft Flight manual Supplement, EAD Aerospace document no.01554-EAS-A-APM-01...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>RAMSEIER, DANIEL</td>
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<td>CTLS-ELA</td>
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<td>Hand-Controls installation on CTLS-ELA</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- To be installed according to Installation Document IN_10050162, Issue2, dated 25 August 2017. To be installed according to the following POH Documents: Document PO_10050162, Issue 1, dated 26 August 2017. To be maintained according to Supplement to Aircraft Maintenance Manual, Document AM_10050162, Issue 2, dated 25 August 2017.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/03/2018</td>
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<td>AIRBUS HELICOPTERS, Inc</td>
<td>FAA STC SH4016SW-D</td>
<td>AS 350 B, B1, B2, B3, BA</td>
<td>Installation of Hourmeter.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Drawing List 350A82-1154, Revision U, dated 25 February 1998, or later FAA approved revision NEV-PAGE</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/03/2018</td>
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<td></td>
<td>A330-203</td>
<td>Cabin Reconfiguration of KLM A330-203/Replacement of Business Class seats, Economy Class seats and Video/Control Centre, installation of monuments, installation of In-Fight/Entertainment System, replacement of decor laminates, curtains and carpet and related changes to cabin furniture to accommodate ne...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, KLM Report MDL-01343S Revision 02, dated 07-Feb-2018 - Instructions for Continued Airworthiness, KLM Report ICA-332-1343S, Revision 09G, dated 05-Jun-2018 or later amendments of the above listed documents approved by EASA.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/02/2018</td>
<td>Active</td>
</tr>
<tr>
<td>10064914</td>
<td>AEROMECH Inc.</td>
<td>10064914</td>
<td>FAA ST01518</td>
<td>E35, E35, E36, E36A</td>
<td>Reduced Vertical Separation Minimum (RVSM)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- AeroTech Document No. AMK-QD-LJ1046 Rev. A, dated May 18, 2006 or latest FAA approved revision - Instructions for Initial and Continued Airworthiness, document MICA-LJ-0428, Rev. A, dated May 18, 2006 or latest FAA approved revision - Aircraft Flight Manual Supplement (AFMS) for the Learjet Model...</td>
<td>- Instructions for Initial and Continued Airworthiness, AMI-ICA-LJ-0428, Rev. A, dated May 18, 2006 or latest FAA approved revision, is a required part of this STC - AMI-AFMS - FAA approved revision, is listed as bestowed.</td>
<td>06/03/2018</td>
<td>Active</td>
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<tr>
<td>10064929</td>
<td>FLIR SYSTEMS AB</td>
<td>10064929</td>
<td>REV. 2</td>
<td>SWEDISH STC M 310/1</td>
<td>AS 350 BB, AS 350 D AS 350 BB, AS 350 C AS 350 D AS 350 B AS 350 B1, AS 350 B2 AS 350 B3, AS 350 BA AS 350 B3 AS 350 BA Installation of External Camera (Fire System AB sensor) Original issue. Transfer of STC to different legal entity from Swedish STC M 310/1 (Rev. 1). Installation of camera PIN 275CFRev. 2: Compliance document revised.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>For original issue PolyTech AB Flight Manual Supplements FNS AS350 No AS550-101, LPV No S41/01 For Rev. 1 Master Document List E200109-841 Rev C, dated 13 June 2021Instructions for Continued Airworthiness E202109-859, Rev B, dated 11June 2...</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point21A.109 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness...</td>
<td>17/06/2023</td>
<td>Active</td>
</tr>
<tr>
<td>10064930</td>
<td>FLIR SYSTEMS AB</td>
<td>10064930</td>
<td>SWEDISH STC M 23D/94</td>
<td>A8206 A, A8206 B 206L-1, 206L-3, 206L-4 206A, 206B, 206L Transfer of the certificate No. M 230/94 Rev. 1 - Installation of PolyTech AB “Stable Eye 350 Type C”, belly or nose mounted.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>PolyTech AB Flight Manual Supplements: LPV No S 1696 Rev 3 (Ba206) and No S11/01 (AB206) or later revisions of the above listed documents approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/07/2018</td>
<td>Active</td>
<td>10064930</td>
</tr>
<tr>
<td>10064931</td>
<td>FLIR SYSTEMS AB</td>
<td>10064931</td>
<td>SWEDISH STC M 24A/94</td>
<td>500N 369HE, 369HM, 369HS 369D, 369F, 369FT, 369H Transfer of the certificate No. M 244/94 Rev. 1 - Installation of PolyTech AB “ARGUS 350 Sensor System” previously named “Stable Eye 350 Type C”, belly or nose mounted.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>PolyTech AB Flight Manual Supplements: LPV No S 5698 belymounted and No S 2196 nose mounted later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/07/2018</td>
<td>Active</td>
<td>10064931</td>
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<tr>
<td>10064932</td>
<td>RAUBECK ENGINEERING Inc.</td>
<td>10064932</td>
<td>FAA STC SA2653VM</td>
<td>B200CT, B200T A200C, A200CT, B200, B200C, B200, 200C, 200CT, 200T, 200T, A200.</td>
<td>Installation of Butterfield Industries Flow Through Anti-Ice Kit PIN B10051-1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Data Package BI 1003, Revision F, dated 08 November 2006 and Drawing list BI 1002, Revision G, dated 04 November 2013 or later FAA approved revisions.</td>
<td>Prior to installation of this modification the installer must determined its the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/06/2018</td>
<td>Active</td>
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<tr>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV_2</td>
<td>LBA 0332657</td>
<td>RALLYE 235 E, RALLYE 235 E-D, MS 894, MS 894 E</td>
<td>Exhaust System MS 235-406500 to Ralleye 235 and MS 894 variants. Extension of the existing cost to following variants Ralleye 235 C, Rallye 235 E, Ralleye 235 F, MS 894 C.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated aircraft design and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>According to GOM Master Document list MO012-204374, rev0_20 180200, To be operated in accordance with Gomolzig AAM-5 Document No. 1240-24-00, Revision 5, dated 2 October 2017, or later FAA approved revision for Continued Airworthiness (ICA) Document No. U160-01490 Revision C, dated 29 September 2017, Aircraft Flight Manual Supplement (AFMS), Document No. U160-0469 2017.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/03/2022</td>
<td>Active</td>
<td>10064933</td>
</tr>
<tr>
<td>U.S. TECHNICAL CONSULTANTS, INC.</td>
<td></td>
<td>FAA STC ST0858A</td>
<td>ATR 72-211A/212A ATR 72-101/202/201/202 ATR 42/200/300/320/400/50 09032018</td>
<td>Installation of two True Blue Power Model TA102 Changing Ports in the cockpit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated aircraft design and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>U.S. Technical Master Data List (MSL), Document No. 7642-24-00, Revision 1, dated 2 October 2017, or later FAA approved revision for Continued Airworthiness (ICA) Document No. U160-0430 Revision C, dated 29 September 2017, Aircraft Flight Manual Supplement (AFMS), Document No. U160-0469 2017.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/03/2018</td>
<td>Active</td>
<td>10064936</td>
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<tr>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td></td>
<td>CAA UK R002559</td>
<td>AS 350 B3</td>
<td>Installation of L3 Trilogy ESI-2000 Standby Instrument, changeHUK/350/1353.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI F-01 Lithium Battery Installation/The requirements for environmental protection and the associated aircraft design and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>AS 365 E301 AFM SUPPLEMENT No. 106AHUK/350-FMS-106, initial issue, dated 10 March 2018 or later revisions of the above listed documents approved by EASA Certification Plan ACP/303135-01 issue C, dated 5 March 2018, Instrucions for Continued Airworthiness ICA/303135-01, issue A, dated 15 February 2018</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/03/2018</td>
<td>Active</td>
<td>10064937</td>
</tr>
<tr>
<td>AIR COMM CORPORATION L.L.C</td>
<td></td>
<td>FAA STC SR06010</td>
<td>MBB-BK117 C-2 MBB-BK117 C-1</td>
<td>Installation of Cabin Air Conditioning System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated aircraft design and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Air Comm Corporation Master Drawing List - EC 145 Cabin Air Conditioning System, Report No. DL-EC145, Revision C, dated 10 April 2006 and FAA approved approved 19 August 2006 or later FAA approved revision 2, Instructions for Continued Airworthiness, Air Comm Corporation Report EC 145-2003-1, Revision 5, dated 09 February 2018</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/03/2018</td>
<td>Active</td>
<td>10064941</td>
</tr>
<tr>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>REV_1</td>
<td>FAA ST01821W1</td>
<td>400A</td>
<td>Installation of the G5000 Integrated Flight Deck, including primary flight displays, automatic flight control system, NAV/COM/FMS and engine instrument systems.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI F-03 (Flight Guidance System) - CRI F-04 (Human Factors) STC 25-ANM-32 (High Altitude Operation at 25,000 feet) - 25-634-1/Airplane Electronic-S...</td>
<td>- Master Drawing List, Garmin International, Inc. document No. 055-02758-00, Revision 1, dated 29 August 2018, Aircraft Flight Manual Supplement, Garmin International, Inc. document No. 190-01508-03, Revision 1, dated 29 August 2018, G5000 System Maintenance Manual, Garmin International, Inc. document...</td>
<td>1. Eligible aircraft serial numbers RK-001 through RK-604. Additional aircraft configuration requirements are defined in the data shown on Master Drawing List (MCL) 005-02756-01), Aircraft modified by this STC must be operated in accordance with Aircraft Flight Manual Supplement (AFMS) identified...</td>
<td>16/03/2018</td>
<td>Active</td>
<td>10064955</td>
</tr>
<tr>
<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA ID</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10064977</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>REV 1</td>
<td>FAA STC ST025796E</td>
<td>A330-231-232/233, A330-201/202/202-223-223F</td>
<td>Interior Reconfiguration on Airbus A330-200 series and A330-300 seriesairplanes.</td>
<td>The Certification Basis for the original product remains applicable to this certificate's approval, except where amended by additional or later amendments if indicated on FAA STC ST025796E. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate's approval.</td>
<td>FAA STC ST025796E</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/11/2018</td>
<td>Active</td>
</tr>
<tr>
<td>10064982</td>
<td>AEROTEC A CONCEPT</td>
<td></td>
<td></td>
<td>737-800</td>
<td>AEROTEC CONCEPT Mod No. 17189 : B737-800 Cabin Reconfiguration.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative arinworthiness requirements:The following paragraph(s) at a later amendment: CS25 Amendment [17]-[25]-7590; The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate's approval.</td>
<td></td>
<td>FAA STC ST025796E</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/03/2018</td>
</tr>
<tr>
<td>10064984</td>
<td>ZEE SYSTEMS, Inc.</td>
<td>FAA STC SAA0345W</td>
<td>500-501</td>
<td>Installation of Zee Systems Air Conditioner.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate's approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate's approval.</td>
<td>FAA STC SAA0345W</td>
<td>FAA Approved Airplane Flight Manual Supplement dated 10 March 1980.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/03/2018</td>
<td>Active</td>
</tr>
<tr>
<td>10064985</td>
<td>AEROLITE AG</td>
<td>REV. 4</td>
<td>EC135 P3, EC135 T3</td>
<td>Installation of Zee Systems Air Conditioner.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative arinworthiness requirements: The following paragraph(s) at a later amendment: CS-27 at Amendment 23a; 27-309(a); 27-312(a); 27-337(a); 27-338; 27-341; 27-351 27-471; 27-619; 27-825(a); ...</td>
<td>FAA STC SAA0345W</td>
<td>FAA STC ST025796E</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/02/2022</td>
<td>Active</td>
</tr>
<tr>
<td>10064986</td>
<td>AIRBUS S.A.S.</td>
<td></td>
<td>A340-643 A340-642</td>
<td>AIS MOD C2093 - A340 GOO26 SYSTEM ACTIVATION/Activation of an Outside Aircraft Equipment (large antenna + radar+shirt) located on A340-600 upper fuselage.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate's approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate's approval.</td>
<td>FAA STC SAA0345W</td>
<td>FAA STC ST025796E</td>
<td>Modification Approval Sheet MAS C2093 Issue 3, dated 16th March 2018 or later revisions of the above listed documents/approved by EASA.</td>
<td>19/03/2018</td>
<td>Active</td>
</tr>
<tr>
<td>10064987</td>
<td>COCETIE AIR FRANCE S.A.</td>
<td></td>
<td>A340-313</td>
<td>&quot;JOON&quot; cabin reconfiguration</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative arinworthiness requirements:25-807, 25-811, 25-853, 25-854, 25-1423 at JAR25 Change 14The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate's approval.</td>
<td>FAA STC SAA0345W</td>
<td>FAA STC ST025796E</td>
<td>Modification Approval Sheet MAS C2093 Issue 3, dated 16th March 2018 or later revisions of the above listed documents/approved by EASA.</td>
<td>15/03/2018</td>
<td>Active</td>
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<tr>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES (DAE)</td>
<td>FALCON TX</td>
<td>Validation of FAA STC ST03591NY. Installation of Honeywell JFL/1ive System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List F7XW- D0042-005 Revision 8, dated 22 January 2018; for Continued Airworthiness F7XW-D0042-156 Revision IR, dated 21 September 2016; Aircraft Flight Manual Supplement F7XW-D0042-150 Revision IR, dated 23 September 2016 for later revisions of the above listed documents.</td>
<td>Prior to installation of this design change it must be determined if this change will introduce any adverse effect upon the airworthiness of the product and must be accepted/rejected on behalf of the aircraft type.</td>
<td>15/02/2018</td>
<td>Active</td>
<td>10064992</td>
<td></td>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>AS 350 B3</td>
<td>Installation of AEM 600W Loud Hailer SystemRev.1.</td>
<td>The Certification Basis for the original product as amended by the following additional or/and alternative airworthiness requirements - The following paragraph(s) at a later amendment 27.773 Amendment ST07-50. The requirements for environmental protection and the associated certificated noise and/or emissions...</td>
<td>- Master Drawing List 300AE3-1794 Rev. A, dated 18 November 2017; Rotorcraft Flight Manual Supplement 300FLM-548 Rev. D dated 16 February 2018.</td>
<td>Prior to installation of this design change it must be determined if this change will introduce any adverse effect upon the airworthiness of the product.</td>
<td>13/04/2021</td>
<td>Active</td>
<td>10064994</td>
<td></td>
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<tr>
<td>HAMILTON SUNDRAND CORPORATION</td>
<td>ST00386BO</td>
<td>Installation of Hamilton Sundstrand Supplemental Cooling Unit (SCU) P/N N701015102EX (service life escalation).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Hamilton Sundstrand Master Document List (MDL) HSNR37062, Revision - dated 30 November 2017, or later FAA-approved revision. Maintenance must be in accordance with Instructions for Continued Airworthiness (ICA), Hamilton Sundstrand Component Maintenance Manual (CMM) 21-54-88, Revision 0, dated 30 November.</td>
<td>Prior to installation of this design change it must be determined if this change will introduce any adverse effect upon the airworthiness of the product.</td>
<td>20/03/2018</td>
<td>Active</td>
<td>10064995</td>
<td></td>
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<tr>
<td>3S CERTIFICATION, LLC</td>
<td>PC-12/47 PC-12/45</td>
<td>Installation of Generator 2 RCCB Upgrade and/or Dual Battery Cockpit Labeling</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. 3S Certification, Master Data List (MDL) 3517-005-1001, Revision IR, dated 17 January 2018 or later FAA-approved revision 2. 3S Certification, Aircraft Flight Manual Supplement (AFMS) 3517-005-7101, Revision IR, dated 17 January 2018 or later FAA-approved revision 3. 3S Certification, Instructions...</td>
<td>EASA validation of FAA STC SA02266LA is a pre-requisite on PC-12/45PC-12/47 for the installation of this STC design change. Prior to installation of this design change it must be determined if this change will introduce any adverse effect upon the airworthiness of the product and/or aircraft.</td>
<td>28/03/2018</td>
<td>Active</td>
<td>10064996</td>
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<tr>
<td>DUNCAN AVIATION, INC.</td>
<td>EMB-135BJ</td>
<td>Upgrade Honeywell FMZ-2000 FMZ to FMZ-2010 for LPV &amp; VNAV/Rev. 1.</td>
<td>The Certification Basis (CB) for the original product as amended by the following additional or/and alternative airworthiness requirements - The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and...</td>
<td>1. STC Master Document List, Duncan Aviation, Inc. Doc. No.CERTDOC-00315, Rev. A, dated 8 November 2017 or later FAA-approved revisions. 2. Aircraft Flight Manual Supplement, Duncan Aviation, Inc. Doc. No.CERTDOC-005289, Rev. A, dated 8 November 2017, or later FAA-approved revisions. 3. Duncan Aviation...</td>
<td>Prior to installation of this design change it must be determined if this change will introduce any adverse effect upon the airworthiness of the product.</td>
<td>16/06/2021</td>
<td>Active</td>
<td>10064997</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>LATITUDE TECHNOLOGIES CORPORATION</td>
<td>DHC-8-401/DHC-8-402/DHC-8-311/DHC-8-314/DHC-8-315/DHC-8-201/DHC-8-202/DHC-8-301/DHC-8-312/DHC-8-103/DHC-8-106</td>
<td>Latitude Technology S200 Satcom and associated accessories.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>- Installation must be in accordance with Latitude Technologies Master Drawing List, 1120-MDL-001, Rev. D, dated 20th of October 2014. Maintenance must be in accordance with Latitude Technologies/inspections for Continued Airworthiness 1120-400-005, Rev. 05.</td>
<td>Prior to installation of this design change it must be determined that the installation will introduce noadverse effect upon the airworthiness of the product.</td>
<td>16/03/2018</td>
<td>Active</td>
<td>10064999</td>
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<tr>
<td>DUNCAN AVIATION, INC.</td>
<td>EMB-135BJ</td>
<td>Installation of a Data Link Communication System supporting future Avionics FANS 1A+</td>
<td>The Certification Basis (CB) for the original product as amended by the following additional or alternative airworthiness requirements: CS25 (amended 17: 25.1317, 25.1317CS25 amended 25: 25.1460) The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. STC Master Document List, Duncan Aviation, Inc. Doc. No.CERTDOC-005598, Rev A, dated 15 November 2017 or later FAA approved revisions. 2. Airplane Flight Manual Supplement, Duncan Aviation, Inc. Doc. No.CERTDOC-005307, Rev B, dated 23 June 2021 or later FAA approved revisions. 3. Duncan Aviation, Inc. Installation Instructions IIN-01/12/2023.</td>
<td>This STC requires that the Honeywell FMZ-2000 Flight Management System/FMS be equipped with software version NZ-6.1.1-03046 (NZ-2010). Prior to installation of this design change it must be determined that the installation will introduce noadverse effect upon the airworthiness of the product.</td>
<td>13/06/2021</td>
<td>Active</td>
<td>10064999</td>
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<tr>
<td>DUNCAN AVIATION, INC.</td>
<td>EMB-135BJ</td>
<td>Upgrade of Honeywell Electronic Display System with option for Synthetic Vision Capability and Primus Elite Avionics.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>STC Master Document List, Duncan Aviation, Inc. Doc. No.CERTDOC-002797, Rev G, dated 30 August 2020.</td>
<td>Prior to installation of this design change it must be determined that the installation will introduce noadverse effect upon the airworthiness of the product.</td>
<td>01/12/2023</td>
<td>Active</td>
<td>10065001</td>
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<tr>
<td>DART AEROSPACE USA, Inc.</td>
<td>MBB-BK117 C-2/ MBB-BK117 D-2</td>
<td>Installation of a Heli-Utility-Basket.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Installation Instructions IN-1121-1044, Revision B, dated 27 October 2017, or later FAA approved revision. Master Drawing List (MDL) MDL-1121-1044, Revision C, dated 11 December 2017, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the installation will introduce noadverse effect upon the airworthiness of the product.</td>
<td>19/03/2018</td>
<td>Active</td>
<td>10065008</td>
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<tr>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY NTER USA, INC.</td>
<td>EMB-505</td>
<td>Installation of a cabin In-Flight Entertainment &amp; Passenger Audio System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>1. Embraer Master Drawing List 16006-001, Revision B, dated 11 September 2016, or later FAA approved revision. 2. Instructions for Continued Airworthiness Document (IAC), Embraer/Engineering &amp; Technology Center (ETC) – USA, Document Number 16006-030, Revision A, dated 4 August 2016, or later approved revision.</td>
<td>Prior to installation of this design change it must be determined that the installation will introduce noadverse effect upon the airworthiness of the product.</td>
<td>22/03/2018</td>
<td>Active</td>
<td>10065014</td>
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<td>STC Holders</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>LATECOERE</td>
<td>A321-272N-271NX A321-273N-271N A321-271N A321-111/-112/-113 A321-115/-116 A321-215/-216 A319-131/-132 A319-111/-112/-113/-114/-115 A318-111/-112/-121/-122</td>
<td>CDSIS installation on Airbus A320 Family. Installation of a Cockpit Door Surveillance System (CDSS) which enables flight crew to identify persons prior to authorizing their entry into the cockpit. Revision 1 is an administrative re-issuance and extends the applicability to A321 long range models.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>CDSIS A320 Family MDL/209AVI-01040-ICA-501- R200073-MMELS-001 LATECOERE MMEL Supplement to AIRBUS Industrie A319/A320 A321 MMEL or later revisions of the above listed document(s) approved under theEASA system.</td>
<td>None.</td>
<td>29/10/2018</td>
<td>Active</td>
<td>10065015</td>
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<tr>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>A340-313</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement AFMS-A340-17-009-001 original issue-Instructions for Continued Airworthiness ICA-A340-17-009-001 original issue or later revisions of the above listed documents approved by EASA. - Change Approval SheetICAAS-A340-17-009 original issue</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness ofthe product.</td>
<td>03/03/2018</td>
<td>Active</td>
<td>10060018</td>
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<tr>
<td>FAA STC SR025925E</td>
<td>R66</td>
<td>NVIS Lighting System</td>
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<td>AVIATION SPECIALTIES UNLIMITED, INC</td>
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<td>20/04/2018</td>
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<td>10060022</td>
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<tr>
<td>AEROSPACE TURBINE ROTABLES, INC d/b/a AEROSPACE TURBINE ROTABLES, JAVOCELS, INC, AERO ELECTRIC, INC</td>
<td></td>
<td>TR-245824 Bural Disk (Large Aerialanes)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Drawing TR245824 Rev. A, dated 30 October 2014-TR-245824-ICA rev C, dated 29 January 2015-EASA-TR-245824-CERT, IR, dated 13 February 2018-TR-245834-COMP, rev N/A, dated 15 April 2015</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness ofthe product.</td>
<td>22/03/2018</td>
<td>Active</td>
<td>10060039</td>
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<tr>
<td>DART AEROSPACE Ltd</td>
<td>S05</td>
<td>Installation of Bearpaws Auxiliary Skid Pad in accordance with MasterDocument List MDL-D050-1049.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Rotorcraft Flight Manual Supplement FMS-D050-1049 Revision A- (&gt;or later revisions of the above listed documents approved by EASA in accordance with the Technical Implementation Procedures of EU-Canada Bilateral Agreement - Master Document List MDL-D050-1049 Revision A- Installation Instructions-...</td>
<td>Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce noadverse effect upon the airworthiness ofthe product.</td>
<td>22/03/2018</td>
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<td>EASA STC No</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC No</td>
<td>Model(s)</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10065053</td>
<td>TELEDYNE BATTERY PRODUCTS</td>
<td>FAA STC 10065053</td>
<td>SR08834DE</td>
<td>MBB-BK117 C-2</td>
<td>Installation of Teledyne Battery Products GIL LT Valve-Regulated-Lithium-Acid (LT-VRLA), FIN 7638-44, battery.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Teledyne Battery Products Document No. TD71-1-01, Master Drawing List, Installation of Teledyne Battery Products P/N 7638-44, Revision IR, dated 19 March 2017, FAA approved 5 April 2017 or later FAA approved revision Instructions for Continued Airworthiness (ICA), Document No. TD 71-8, 8C. Revision NC, d...</td>
<td>Prior to installation of this certification change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/03/2018</td>
<td>Active</td>
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<td>10065057</td>
<td>AEROSPACE TURBINE ROTABLES, INC db/KELLY AEROSPACE TURBINE ROTABLES, IAVCELLS, INC, AER ELECTRIC, INC</td>
<td>441</td>
<td>425</td>
<td>TR-245824 Burn Disk (General Aviation)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23.1411(c)(3)(b), 25.1051(a)(1)(ii)(b), 25.1031(a)(2)(iii), 25.1201(a)(iii), 25.1301(a)(ii)(c)(iv) These requirements for environmental protection and the associated...</td>
<td>-Drawing TR245824 Rev. A, dated 30 October 2014: TR-245824/ICA revision C, dated 29 January 2015, EASA-TR-245824-CERT, IR, dated 13 February 2018-TR-245824-COMP, rev N/A, dated 15 April 2015</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/03/2018</td>
<td>Active</td>
<td>10065057</td>
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<tr>
<td>10065061</td>
<td>SCANDINAVIAN AVIONICS DESIGN A/S</td>
<td>AB412 EP</td>
<td>NVG compatible cockpit lighting and NVG friendly exterior lighting modification, ref. 21138601.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>2103601-AM A, FLIGHT MANUAL SUPPLEMENT, Issue 2, dated 06 February 2018(+) or/and later revisions of the above listed document(s) approved under the EASA system.</td>
<td>Prior to installation of this change repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/03/2018</td>
<td>Active</td>
<td>10065061</td>
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<tr>
<td>10065066</td>
<td>FLYING COLOURS CORP.</td>
<td>TCCA C-LSA17-119D</td>
<td>CL-600-2B16 VARIANT</td>
<td>Interior Completion with Provisions For Air Ambulance Equipment.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 23.1610(a)(ii)(b), 25.101(c)(2)(ii)(a)(ii)(c)(iii), 25.1201(a)(iii), 25.1301(a)(ii)(c)(iv) These requirements for environmental protection and the associated...</td>
<td>- Modification Data Summary K604000 Rev B dated 13 Mar 2018: Mandatory Maintenance Manual Supplement K604075 Rev. A dated 11 March 2018; or later modifications of the above listed document(s) approved on behalf of EASA in accordance with the T...</td>
<td>Prior to installation of this design change the interrelationship between this change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/03/2018</td>
<td>Active</td>
<td>10065066</td>
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<tr>
<td>10065073</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC 10065073</td>
<td>SH43453WV-D</td>
<td>AS 350 B3 AS 350 B1/AS 350 B2 AS 380 B3/AS 350 BA AS 350 D</td>
<td>Installation of strobe anti-collision light installation and/ or supplemental strobe light.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Drawing List 350AS-1176, Revision Y, dated 25 April 1998, or later FAA Approved revision Drawing List 350AS-1176, Revision AP, dated 9 June 2015(+), or later CODA approved revision Instructions for Continued Airworthiness (ICA) 35001MS-C00, Revision 0(+), dated 29 June 2015(+), or later FTW.</td>
<td>Installation of LED strobe anti-collision light for AS-350B2 and AS-350B3 models only. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect...</td>
<td>26/03/2018</td>
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<tr>
<td>10065076</td>
<td>MAGELLAN AEROSPACE</td>
<td>TCCA STC 10065076</td>
<td>SH17-21 ISSUE 1</td>
<td>505</td>
<td>Installation of Wire Strike Protection System in accordance with Modification Data Summary EM107462.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>-Modification Data Summary EM107462 Issue B - Instructions for Continued Airworthiness and Illustrated Parts ListPM-965-094 Issue C</td>
<td>Prior to installation of this change repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/03/2018</td>
<td>Active</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AS 332 L AS 332 L AS 332 L AS 332 L AS 332 L</td>
<td>Installation of GARMIN GTX Transponder System/This change covers the installation of a GARMIN GTX320/ GTX340/8R Transponder, controlled by a Garmin GTN750. The installation of the GTN50 is not part of this installation Rev. 1: Transponder Model deleted from title and description, CS ACNS issue 3 instead...</td>
<td>The Certification Basis for the original product as amended by the foregoing additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment CS ACNS issue 3 CS ACNS D ELS 001, CS ACNS D ELS 019, CS ACNS D ELS 019 (Y) CS ACNS D ELS 020, CS ACNS D ELS 025 CS ACN...</td>
<td>Flight Manual Supplement, Doc. No. HAU-D-038-005, Rev. 1: Master Draw List, Doc. No. HAU-D-038-001, Rev. 1 Installation Manual, Doc. No. HAU-D-038-004, Rev. 1: Instructions for Continued Airworthiness, HAU-D-038-008, Rev. 1: or later revisions of the above listed document(s) approved/ accepted under the...</td>
<td>Required prerequisite for this change is the STC HAU-D-029 or a similar installation introducing a GTN50(X) or GTN750(X). Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce...</td>
<td>22/03/2022</td>
<td>Active</td>
<td>10065075</td>
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<tr>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>EMS interior P/N 650-0212923-501</td>
<td>The Certification Basis for the original product as amended by the foregoing additional or alternative airworthiness requirements. Special Condition (y) CRI D-01 Medical Evacuation Delegation(s). CRI D-02 Width of Asia/The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Draw List MDL-15-017 Rev. 0 ctd. 2018.03.26 - Fight Manual Supplement FMS-650-0212923-501 Rev. 0 ctd. Mar. 22/2018 - Operations Manual OPM-650-0212923-501 Rev. 0 ctd. Mar. 12/2018 for later revisions of the above listed document(s) approved under theEASA system.</td>
<td>- Fight Manual Supplement FMS-650-0212923-501 - Operations Manual OPM-650-0212923-501 Pre requisite: EASA STC 10065066 (validated TCCA STC C-LSA17-19 D)/ Interior Completion with Provisions for Air Ambulance Equipment&quot; - Limited to serial No. 6092, 6105, 6114 For installation of this change/repair...</td>
<td>27/03/2018</td>
<td>Active</td>
<td>10065064</td>
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<td>FAA STC ST0255SE</td>
<td>Installation of an interior reconfiguration on Boeing model 777-300ER series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Jamco America, Inc. Master Draw List (MDL) MDL2295-1 Revision B dated 21 February 2018, or later Federal Aviation Administration (FAA) approved revision Supplemental Instructions for Continued Airworthiness (ICA) ICA22950-1 Revision 1 dated 12/2018 and Supplemental Electrical Wiring...</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise and/or vibration effect upon the airworthiness of the product.</td>
<td>28/03/2018</td>
<td>Active</td>
<td>10065060</td>
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<tr>
<td>FAA STC ST0181SE-D</td>
<td>Installation of an interior reconfiguration on Boeing model 777-300ER series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Jamco America, Inc. Master Draw List (MDL) MDL2295-1 Revision B dated 21 February 2018, or later Federal Aviation Administration (FAA) approved revision Supplemental Instructions for Continued Airworthiness (ICA) ICA22950-1 Revision 1 dated 12/2018 and Supplemental Electrical Wiring...</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise and/or vibration effect upon the airworthiness of the product. This modification may not be installed after 7...</td>
<td>04/07/2018</td>
<td>Active</td>
<td>10065094</td>
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<tr>
<td>FAA STC ST0119SE</td>
<td>Installation of Door 1 and Door 2 interior modification on Boeing Model 777-300ER series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Jamco America, Inc. Master Draw List (MDL) MDL1847-1 Revision A dated 7 December 2015, or later Federal Aviation Administration (FAA) approved revision Supplemental Instructions for Continued Airworthiness (ICA) Document Number ICA1847-1 Revision 0 dated 13 November 2015 and Supplemental Electrical Wiring...</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise and/or vibration effect upon the airworthiness of the product.</td>
<td>27/03/2018</td>
<td>Active</td>
<td>10065095</td>
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<td>STC No.</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10065104</td>
<td>LEARJET INC.</td>
<td>LEARJET MODEL 45</td>
<td>Installation of Honeywell Ov-875 Primus Elite cockpit display units in the Learjet Model 40/45.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>According to LEARJET Inc., Master Document List BAG0417005, revision (c) dated July 14, 2017 FAA approved July 18, 2017. To be operated in accordance with LEARJET Inc., AFM-S Document no current approved manual. To be installed i.e. Master Document List BAG0417005, revision (c) dated July 14, 2017 FAA..</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/04/2018</td>
<td>Active</td>
<td>10065104</td>
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<td>10065107</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A340-313</td>
<td>VIP Conversion &quot;Silence&quot; Project</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CRI A-101 &quot;Certification Basis.&quot; The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product are unchanged.</td>
<td>- Supplemental Type Definition Document, Ref. STD000-16/23M Issue 01 Rev.00. CMA. Ref. A-00-17/13-ICA Issue 01 Rev. 00 (Section 4) AFM. Ref. A-00-17/13-AM Issue 01 Rev. 00 (later revisions of the above listed document(s) approved under the EASA system. - Compliance Record List, Ref. CRL00-16/2M Issue..</td>
<td>As defined in the Airplane Flight Manual Supplement, Ref. A-00-17/13-AM (Section 2), and the Instructions for Continued Airworthiness, Ref. A-00-17/13-ICA (Section 4). Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other...</td>
<td>28/03/2018</td>
<td>Active</td>
<td>10065107</td>
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<td>10065108</td>
<td>DIAMOND AIRCRAFT INDUSTRIES GMBH</td>
<td>DA 62</td>
<td>DA 62 MPP Basic Options/Revision &quot;n&quot; of STC 62-001 certifies BR-700 Pod – Module 185/Revision &quot;a&quot; certifies Maximum Take-Off Mass 2360 kg with PPC 185 Revision &quot;aa&quot; certifies Maximum Take-Off Mass 2360 kg.</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements CRI A-101 &quot;Certification Basis.&quot; The requirements for environmental protection and the associated/required noise and/or emissions levels of the original product are unchanged.</td>
<td>Installation per Supplemental AMs: Doc. No. 7.02.25-M00 Rev. 00. Doc. No. 7.02.25-M20 Rev. 00. Doc. No. 7.02.25-M30 Rev. 00. Doc. No. 7.02.25-M50 Rev. 00. Doc. No. 7.02.25-M70 Rev. 00. Doc. No. 7.02.25-M80 Rev. 00. Doc. No. 7.02.25-M100 Rev. 00. Doc. No. 7.02.25-M150 Rev. 00. Doc. No. 7.02.25-M180 Rev. 00. Operation according to...</td>
<td>Prior to installation of this change design change it must be determined that the interrelationship between this change design and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/05/2024</td>
<td>Active</td>
<td>10065108</td>
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<tr>
<td>10065112</td>
<td>BERINGER AERO</td>
<td>PA-46R-350T, PA-46-350P, PA-46R-350T, PA-46-350TP, PA-46-350P, PA-46R-350T, PA-46-350P, PA-46-301P, PA-46-300TP, PA-46-300P, PA-46-301P, PA-46-310P, PA-46-310P</td>
<td>Servicing instruction &amp; SamsAIR option/Revision 2 of this STC introduces an adapter for the bar;bar/Revision 5 is an administrative change and addition of SamsAIR option/Weight correction/Revision 4 accounts for correction and update of the CAA-STC-017, revision 4.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval...</td>
<td>- Installation and Instructions for Continued Airworthiness: ICA- STC-017 rev 5 dated 22 May 2023. - Project specific Certification Plan: DM: MOD002-STC-017 rev 00 dated 17 February 2022 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/08/2023</td>
<td>Active</td>
<td>10065112</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>04/05/2018</td>
<td></td>
<td>750</td>
<td>Installation of modifications for introduce A2S-B Out capability intoCessna Citation X aircraft equipped with Primus II Radio Suites.</td>
<td>The Certification Basis for the original product as amended bythe following additional or alternative airworthiness requirements: Refer to the CRI A-01 for EASA STC 1006511. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable.</td>
<td>- Supplemental Type Design Document STC003-176/16/02 Issue 04; Revision 01, Instructions for Continuous Airworthiness, Rev. A-00-180/16A01/16A02/16A03 Issue 03 (Chapter A / Airworthiness Limitation Section) or later revisions of the above listed document(s) approved under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>04/05/2018</td>
<td>Active</td>
<td>10065118</td>
</tr>
<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>29/03/2018</td>
<td></td>
<td>A319-312</td>
<td>Installation of a tracking system GTS V2, transmitting the aircraft's position over the Inmarsat satellites Rev. 1, New GTS V2 amendment and modification of the supporting bracket.</td>
<td>The Type and OSD Certification Bases (CB) for the original product are unchanged and remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- “STC Document List For GNSS on B737”, Issue 29 dated 29/03/2024</td>
<td>Embedment of STC 10066133 is pre-requisite for this change. The approval holder shall fulfill the obligations of Part 21, Point 21.A.109 Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change shall be assessed together with the installation of this change/repair.</td>
<td>27/03/2024</td>
<td>Active</td>
<td>10066133</td>
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<tr>
<td>AIRBUS S.A.S.</td>
<td>05/04/2018</td>
<td></td>
<td>A380-841, A380-842, A380-851</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements Equivalent Safety Findings (ESF) C-02 Vibration and Buffeting/Aerodynamic Stability Requirements Compliance Criteria The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable.</td>
<td>- NAS-C.10.11/G2.11 Issue 01. MPD- Supplement ref. MPD-5-AS04 MDD112148 Rev 06</td>
<td>None</td>
<td>05/04/2018</td>
<td>Active</td>
<td>10066134</td>
<td></td>
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</tbody>
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**Notes:**
- STC: Supplemental Type Certification
- Each STC has specific requirements and limitations that must be followed during installation.
- Prior to installation, the approval holder must ensure that all requirements are met and that the installation does not introduce any adverse affects on the airworthiness of the product.
- The STCs are subject to revision and updates, and it is important to check for the most current information.
- EASA: European Union Agency for Safety of Aviation
- MPD: Modular Procedure Document
- STDD: Supplemental Type Design Document
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<th>STC Holder</th>
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<th>Certification basis</th>
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<tr>
<td>AVIONIK STRAUßING ENTWICKLUNGS GmbH</td>
<td>FAA STC SA02658LA</td>
<td>B300C B300C-200, B300C-300, 300LW B300, B300C-200, B300C-300, 300LW</td>
<td>Installation of Garmin GAVS and/or L-3 Communications NY-164 antenna(s)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Avionik Straubing modification reference ASR-2017-002 as defined in MOL ref ASR-2017-002-MOD-01-00 dated 03.03.2018 and Instructions for Installation (IF) dated 03.03.2018</td>
<td>Prior to installation of this change/repair it may be determined that the modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/04/2018</td>
<td>Active</td>
<td>10065153</td>
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<tr>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA STC SA02658LA</td>
<td>B300C B300C-200, B300C-300, 300LW B300, B300C-200, B300C-300, 300LW</td>
<td>Replacement of the original Pratt and Whitney PT6A-60A engines and propellers with Pratt and Whitney PT6A-67A engines and MT Propeller M2-27-1 N-C F-R/J C/F 2G 650 650 propellers or Hartzell HC-E-ESA A0100458 propellers</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA02658LA. The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List 200914-00, Rev. A dated 23 May 2017, or later FAA approved revisions. Pilot’s Operating Handbook and Airplane Flight Manual Supplement AFMS200914-20. Rev. IR dated 14 August 2017, or later FAA approved revisions required. The following Pilot’s Operating Handbook and FAA approved revision:</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/11/2020</td>
<td>Active</td>
<td>10065145</td>
</tr>
<tr>
<td>AEROSPACE TURBINE ROTABLES, INC. d/b/a KELLY AEROSPACE</td>
<td>FAA STC SA02658LA</td>
<td>S-76A, S-76B, S-76C 212, 412, 412EP</td>
<td>TR-245824 Bural Disk - CS-29 STC approval of FAA-PMA approved critical PMA part.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Drawing TR245824 Rev. A, dated 30 October 2014, TR-245824-ICA revision C, dated 29 January 2015. EASA-TR-245824-CERT, IR dated 13 February 2018; TR-245824-COMP, rev N/A, dated 15 April 2015.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/04/2018</td>
<td>Active</td>
<td>10065149</td>
</tr>
<tr>
<td>VENTURA AEROSPACE, Inc.</td>
<td>FAA STC ST0967LA</td>
<td>737-800</td>
<td>Installation of 9g Rigid Cargo Barrier.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Ventura Aerospace, Inc. Master Data List VA-MDL-1066, Revision A, dated 13 March 2018 or later FAA approved revision. 2. 737-800 Boeing Converted Freighter (BCF) (Flying Top Drawing number 800A0003, Revision C, dated 27 March 2018, or later FAA approval revision(s). 3. Instructions for Continued Airworthiness.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/04/2018</td>
<td>Active</td>
<td>10065167</td>
</tr>
<tr>
<td>ANORA INTERNATIONAL LLC</td>
<td>FAA STC ST05966LA</td>
<td>737-800</td>
<td>Installation of main deck cargo loading system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Ancora International LLC Controlled Data List DL6560, Revision C, dated December 2017, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/04/2018</td>
<td>Active</td>
<td>10065171</td>
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List of EASA Supplemental Type Certificates

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<tr>
<td>ANDERSON, THOMAS RICHARD</td>
<td></td>
<td>FAA STC SA00805E</td>
<td></td>
<td>Installation of Steel Main Landing Gear.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any previously installed design change and/or repair will introduce a misuse effect upon the airworthiness of the product. The STC is valid for those aircraft types and model...</td>
<td>10/04/2018</td>
<td>Active</td>
<td>1005517A</td>
</tr>
<tr>
<td>ANDERSON, THOMAS RICHARD</td>
<td></td>
<td>FAA STC SA2359NM</td>
<td></td>
<td>Installation of tail wheel assembly.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any previously installed design change and/or repair will introduce a misuse effect upon the airworthiness of the product. The STC is valid for those aircraft types and model...</td>
<td>10/04/2018</td>
<td>Active</td>
<td>1005517F</td>
</tr>
<tr>
<td>FINNOFF AVIATION PRODUCTS, LLC</td>
<td></td>
<td>FAA SA02264LA</td>
<td></td>
<td>Installation of a Pratt and Whitney PT6A-67P Engine Upgrade/whitsopthical 2nd Battery Installation and optional Generator/2 Exhaust Ducting installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Master Drawing List (MDL) No. 0201-00100, Revision A, dated 11th November 2011; for operation in accordance with AFM SuppFta- FIN-001022, Revision A, dated 18th February 2012 or later revision(s).</td>
<td>10/04/2018</td>
<td>Active</td>
<td>1005517H</td>
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<tr>
<td>ARMSTRONG AEROSPACE Inc. dba ASTRONICS ARMSTRONG AEROSPACE</td>
<td></td>
<td>FAA ST0321CH</td>
<td></td>
<td>Installation of Additional Steel Main Landing Gear.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: EASA CS 25.1309(a) and/or CS 25.1529; if applicable: The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Armstrong Aerospace Master Data List, Doc. No. OC104-9012-00, Revision E, dated September 26th, 2016 or later approved revisions.</td>
<td>10/04/2018</td>
<td>Active</td>
<td>1005518E</td>
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<tr>
<td>STC TWENTY ONE LIMITED</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>757-200F</td>
<td>Introduction Of An Underwater Locator Beacon (ULB).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (a): ORI F-01: Non-rechargeable Lithium Battery installations. - The following paragraph(s) at a later amendment: CS 25 Amendment 20; CS25.601; CS25.603; CS...</td>
<td>STC TWENTY ONE MASTER DOCUMENT LIST No. S21: MDL-1383 ISSUE1.</td>
<td>10/04/2018</td>
<td>Active</td>
<td>10055200</td>
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<tr>
<td>WILLIBALD LANG</td>
<td></td>
<td>LS 3-A</td>
<td></td>
<td>Bau und Anbringung von Winglets Das Maßnahme umfasst den Bau von Winglet in Formen, sowie das anbringender Winglets an den Flächenenden der LS3-a.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment: JAR 22.375 Winglets: The requirements for environmental protection and the associated certificated noise and/or emissions levels of the...</td>
<td>Zeichnung: Winglet (LS3-a) Li / Re Nr.: LS3-1311 Li -1312/01/00 Compliance Checklist LS3-a vom 13.03.2018/Erweiterungen zum Flughandbuch vom 01.02.1995/der later revisions of the above listed document(s) approved under theEASA system. Only for Sailplane LS3-a, Werk-Nr.: 3183.Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any previously installed change and/or repair will introduce a misuse effect upon the airworthiness of the product.</td>
<td>12/04/2018</td>
<td>Active</td>
<td>1005521A</td>
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10065220  
**STC Holder**: AIRBUS HELICOPTERS, Inc.  
**Rev. level**: REV. 3  
**Foreign/NA AIC**: SR09500RC-D  
**Model**: EC 130 B4, EC 130 T2 AS 350 B2, AS 350 B3  
**Description**: Installation of a L-3 Trilogy ES-3000 or ESI-500 Electronic Standby Indicator Rev. 3 adds an optional configuration with new ICA.  
**Certification basis**: The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.  
**Associated Technical Documentation**: Airbus has renumbered the drawing list 35A83-1791 Revision B, dated 7 February 2014. Certification (CB) A350-1000/-900 models:Rotorcraft Flight Manual Supplement (RFM) 35FMS-528 Revision 1 (dated 6 March 2014). Instructions for Continued Airworthiness (ICA) 35FMS-028 Revision 1 (dated 12 Feb...  
**Limitations/Conditions**:  
**Date of Issue**: 21/01/2022  
**Status**: Active  
**System Record**: 10065223

10065222  
**STC Holder**: PIAV ENGINEERING  
**Rev. level**: REV. 1  
**Model**: 737-800/900/900ER  
**Description**: The STC consists on the installation of the ADS-B OUT function on B737 NG.  
**Certification basis**: The Certification Basis for the original product as amended for the following paragraph(s) at a later amendment: CS25 adds 4.25.1301(a)(b)(c), 25.1302(b), 25.1303(a)(b)(c) and 25.1302(g)(v)(w)  
**Associated Technical Documentation**:  
**Limitations/Conditions**:  
**Date of Issue**: 29/06/2019  
**Status**: Active  
**System Record**: 10065223

10065226  
**STC Holder**: COMTECH GLOBAL AVIATION SERVICES S.  
**Rev. level**: REV. 1  
**Description**: Introduction of Centre Fuel Tank Ignition Mitigation Reticulated Polyurethane Foam.  
**Certification basis**: The Certification Basis for the original product as amended for the following paragraph(s) at a later amendment: CS25 adds 25.301(a)(b)(c), 25.302, 25.307(a), 25.321(a), 25.333, 25.343(a), 25.401, 25.402, 25.406, 25.609, 25.611, 25.901  
**Limitations/Conditions**: Prior to installation of this change/repair it must be determined that the system remains or becomes airworthy within the limits of the certificate/approval.  
**Date of Issue**: 04/04/2019  
**Status**: Active  
**System Record**: 10065223

10065227  
**STC Holder**: ICELANDAIR EHF  
**Rev. level**: REV. 1  
**Model**: 737-8, 737-9  
**Description**: New seating layout with 160 total passengers using AVIO seats with 2D/3D IFE and ISPS systems.  
**Certification basis**: The Certification Basis for the original product as amended for the following paragraph(s) at a later amendment: CS25 adds 1.29(a)(b), 301(a), 302, 303, 307(a), 308(g), 510(a)(b)(c), 582(u)(w)(v), 571, 601, 603(a)(b)(c), 605(a)(b)  
**Limitations/Conditions**: Prior to installation of this change/repair it must be determined that the system remains or becomes airworthy within the limits of the certificate/approval.  
**Date of Issue**: 16/06/2019  
**Status**: Active  
**System Record**: 10065223

10065229  
**STC Holder**: CORPORATION LOGIC AIR Inc.  
**Rev. level**: REV. 1  
**Model**: 737-500  
**Description**: Structural Provisions for Dual GPSInstallation of structural provisions for dual GPS antennas on B737-500/500 aircraft.  
**Certification basis**: The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.  
**Associated Technical Documentation**: -K388-102-001-002 Rev. B “Modification Data Summary” -K388-102-001-012 Rev. B “ Mandatory Maintenance Manual Supplement” or later revisions of the above listed document(s) approved on behalf of EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement.  
**Limitations/Conditions**: After endorsement of this STC, the aircraft must be maintained in accordance with the applicable instructions for continued airworthiness/report K388-102-001-012 Prior to installation of this change/repair it must be determined that the system remains or becomes airworthy within the limits of the certificate/approval.  
**Date of Issue**: 30/04/2018  
**Status**: Active  
**System Record**: 10065223
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>NSE</td>
<td>A321-253N-272N</td>
<td>KA-BAND Installation on Airbus SATs: Major Change to STC 10065237 Rev 1 (issued 28 Nov 2019) consists in adding a new design change configuration by installing the KA-BAND satellite antenna on existing SBB antenna doubler. This design change requires small adaptation of the installation.</td>
<td>The Certification Basis for the original product as amended by the preceding additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CR-B-0301-002 (Vibration/ buffeting/compliance criteria for large external antenna installation) [The requirements for environmental protection...</td>
<td>Master Document List 1106246</td>
<td>2020-06-29</td>
<td>Active</td>
<td>10065237</td>
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<tr>
<td>EAD AEROSPACE</td>
<td>A321-253N-272N</td>
<td>Certification Basis for the original product as amended by the preceding additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CR-B-0301-002 (Vibration/buffeting/compliance criteria for large external antenna installation) [The requirements for environmental protection...</td>
<td>Master Document List 1106246</td>
<td>2020-06-29</td>
<td>Active</td>
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<tr>
<td>AEROWORKINESS</td>
<td>A321-253N-272N</td>
<td>The Certification Basis for the original product as amended by the preceding additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CR-B-0301-002 (Vibration/buffeting/compliance criteria for large external antenna installation) [The requirements for environmental protection...</td>
<td>Master Document List 1106246</td>
<td>2020-06-29</td>
<td>Active</td>
<td>10065237</td>
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<tr>
<td>CONCORDE BATTERY CORPORATION</td>
<td>A321-253N-272N</td>
<td>The Certification Basis for the original product as amended by the preceding additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CR-B-0301-002 (Vibration/buffeting/compliance criteria for large external antenna installation) [The requirements for environmental protection...</td>
<td>Master Document List 1106246</td>
<td>2020-06-29</td>
<td>Active</td>
<td>10065237</td>
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<tr>
<td>FAAC ST TAD2653A</td>
<td>510</td>
<td>The Certification Basis for the original product as amended by the preceding additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CR-B-0301-002 (Vibration/buffeting/compliance criteria for large external antenna installation) [The requirements for environmental protection...</td>
<td>Master Document List 1106246</td>
<td>2020-06-29</td>
<td>Active</td>
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<tr>
<td>JET AVIATION AG</td>
<td>767-300</td>
<td>The Certification Basis for the original product as amended by the preceding additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CR-B-0301-002 (Vibration/buffeting/compliance criteria for large external antenna installation) [The requirements for environmental protection...</td>
<td>Master Document List 1106246</td>
<td>2020-06-29</td>
<td>Active</td>
<td>10065237</td>
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<tr>
<td>DASSAULT FALCON JET</td>
<td>Falcon 2000EX</td>
<td>Certification Basis for the original product as amended by the preceding additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CR-B-0301-002 (Vibration/buffeting/compliance criteria for large external antenna installation) [The requirements for environmental protection...</td>
<td>Master Document List 1106246</td>
<td>2020-06-29</td>
<td>Active</td>
<td>10065237</td>
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### List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC ST105745C-D</td>
<td>FALCON 200EX</td>
<td>Aircraft Compliance for Falcon 2000EX S/N 115</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Data List Report No. F2EX-07011-D1019 revision NC dated 7 May 2007, or later FAA approved revision. 2. FAA Approved AFM Supplement, Report 37159- F2EX-0109 dated 8 May 2901 or later FAA approved revision. 3. Supplemental Maintenance Manual, Report SMM F2000EX- SN115 dated April 2007 or later FAA a.</td>
<td>You cannot use descriptive data about this design change to dupulicate other products. This approval is limited to the installation made in Dassault Aviation Model Falcon 2000EX Serial No. 115. This STC does not permit manufacturing of parts for multiple installations. Prior to installation of this change/repair.</td>
<td>19/04/2018</td>
<td>Active</td>
<td>10065285</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC ST105729C-D</td>
<td>FALCON 200EX</td>
<td>Aircraft Compliance for Falcon 2000EX S/N 91</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Data List Report No. F2EX-07011-D991, Revision A dated 14 December 2006, or later FAA approved revision. 2. FAA Approved AFM Supplement, Report 37159- F2EX-0901 dated 15 December 2006 or later FAA approved revision. 3. Supplemental Maintenance Manual, Report SMM F2000EX- SN 91, Revision C dated December 2006...</td>
<td>You cannot use descriptive data about this design change to dupulicate other products. This approval is limited to only the installation made in Dassault Aviation Model Falcon 2000EX S/N 91. This STC does not permit manufacturing of parts for multiple installations. Prior to installation of this change/repair.</td>
<td>19/04/2018</td>
<td>Active</td>
<td>10065287</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC ST105729C-D</td>
<td>FALCON 200EX</td>
<td>Installation of ST3100 Satellite Telephone System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Data List Report No. F2EX-02017, Revision NC dated 8 December 2006, or later FAA approved revision. 2. FAA Approved AFM Supplement, Report 37786-1 dated 15 December 2006 or later FAA approved revision. 3. Maintenance Manual Supplement Report 37549, Revision D dated 8 December 2006 or later FAA approved.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/04/2018</td>
<td>Active</td>
<td>10065288</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC ST105729C-D</td>
<td>FALCON 200EX</td>
<td>Installation of Service and Indirect Lighting System for Falcon 2000EX</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Data List Report No. F2EX-02019 Revision B dated 20 June 2005, or later FAA approved revision. 2. Supplemental Maintenance Manual Supplement Report 37490 dated 17 May 2005 or later approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/04/2018</td>
<td>Active</td>
<td>10065292</td>
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<tr>
<td>LUFFTHANSA TECHNIN AG</td>
<td>REV. 1</td>
<td>A340-313</td>
<td>Intrusion Detection System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>- Supplemental Type Design Document, Ref. STD201-1701M, Issue 2, Rev 0, dated 08 July 2019. - Aircraft Flight Manual Supplement, Ref. A-00-17/80-FAFM, Issue 1, Rev 1, dated 21 March 2019. Instruction for Continued Airworthiness, Ref. A-00-17/80-ICA, Issue 3, Rev 0, dated 26 June 2019 later revision.</td>
<td>- The EASA STC 10033112 (VIP Cabin Conversion) must be installed. As per ICA, Ref. A-00-17/80-ICA, Section 4. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/10/2019</td>
<td>Active</td>
<td>10066224</td>
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*EASA STC No.*

*STC Holder*

*Rev. level*

*Foreign/NAA STC ref.*

*Model*

*Description*

*Certification basis*

*Associated Technical Documentation*

*Limitations/Conditions*

*Date of Issue*

*Status*

*System Record*
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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA code</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<th>Date of Issue</th>
<th>Status</th>
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<tbody>
<tr>
<td>10065295</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC ST104129C-D</td>
<td>FALCON 200EX</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Data List Report No. F2EX-0701-D027H (&lt;&gt;) Revision C dated 27 July 2005, or later FAA approved revision 2. FAA Approved AFM Supplement, Report 37159: F2EX-0259 dated 2 August 2005.3. Supplemental Maintenance Manual, Report SMIM F2000EX-SN 059 dated July 2005.</td>
<td>Descriptive data pertaining to this design change are consistent and adequate for duplication on other products. This approval is limited to only the installation made in Dassault Aviation Model Falcon 2000EX S/N 59. This STC does not permit manufacturing of parts for multiple installations. Prior to installation.</td>
<td>19/04/2018</td>
<td>Active</td>
<td>10065295</td>
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<tr>
<td>10065296</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC ST105075C-D</td>
<td>FALCON 200EX</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Data List Report No. F2EX-0701-D029H (&lt;&gt;) Revision C dated 21 August 2006, or later FAA approved revision 2. FAA Approved AFM Supplement, Report 37159: F2EX-0258 dated 22 August 2006, or later FAA approved revision 3. Supplemental Maintenance Manual, Report F2000EX-SN 060 dated August 2006, or later.</td>
<td>You cannot use descriptive data about this design change to duplicate other products. This approval is limited to only the installation made in Dassault Aviation Model Falcon 2000EX S/N 86. This STC does not permit manufacturing of parts for multiple installations. Prior to installation.</td>
<td>19/04/2018</td>
<td>Active</td>
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<tr>
<td>10065303</td>
<td>ICELANDAIR DESIGN ORGANISATION</td>
<td>FAA STC ST003260B</td>
<td>767-300 SERIES</td>
<td>Installation of Thales Integrated Electronic Stand-by Instrument (IESI) V2 Family 2) P/N C16786AAA02.</td>
<td>1. Icelandair Master Document List (MDL) Document numberC9A-1266-345Q-012, Revision D dated 27 March 2018, or later FAA approved revision 2. Icelandair Aircraft Flight Manual Supplement (AFMS) Document numberI266-347Q-012, Revision A, FAA approved on 7 March 2018, or later FAA approved revision 3...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/04/2018</td>
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<td>10065303</td>
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<td>10065308</td>
<td>328 SUPPORT SERVICES GmbH</td>
<td>HAWKER 800XP</td>
<td>HAWKER 850XP</td>
<td>Installation according to Master Document List: MLA-2006 Document code 55-A, dated 19/04/2018 Operation according to Aircraft Flight Manual Supplement 12002231 20061 Issue A, dated 20061 Issue A, dated 04/04/2008. Maintenance according to Aircraft Maintenance Manual Supplement AMS-02231 20061 Issue A, dated 04/04/2008.</td>
<td>Limited to aircraft with Rockwell Collins Proline 21 avionics as well as all other avionics. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the aircraft...</td>
<td>23/04/2018</td>
<td>Active</td>
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<tr>
<td>10065309</td>
<td>WEST STAR AVIATION, L.L.C.</td>
<td>FAA STC ST04120CH</td>
<td>MYSTERE FALCON 900 (MF900)</td>
<td>Installation of a Datalink Communications System supporting IFR Traffic Services (ATS) using Honeywell MK-III Communications Management Unit (CMU), Honeywell TR-668 VHF DataLink radio, Aircell Axxess IridiumSatcom, and a GoGo Iridium Data Interface Unit (DIU).</td>
<td>1. West Star Aviation Master Data List Document No. DB1504002 Revision 2, dated 11 July 2016, or later FAA approved revision 2. Aircell Flight Manual Supplement (AFMS) DB1504005 Revision 1, dated 16 August 2016, or later FAA approved revision</td>
<td>The aircraft CVR must be capable of supporting CPDLC data link recording. Prior to installation of this change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft...</td>
<td>25/04/2018</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>GFD GmbH</td>
<td>Iridium Satcom Antenna Installation Provision Rev. 1: Transfer to GFD GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this change/repair it must be determined whether the interconnectedness between the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/02/2022</td>
<td>Active</td>
<td>10065314</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>Zonal Drying System Installation Revision 1: extended applicability</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraphs (s) at a later amendment: EASA CS 25.0259.25.0301, 25.0305, 25.0307, 25.0365, 25.0571, 25.0581, 25.0601.25.0603, 25.0609, 25.0...</td>
<td>Supplemental Type Design Definition STOD21-1701M Issue 2: Instr. Requirements for Cont. Airworthiness A-00-18/32-ICA issue 5 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>12/01/2023</td>
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<td>DASSAULT FALCON JET CORPORATION</td>
<td>Upgrade Honeywell FMZ-2000 Fight Management System (FMS) to Software Version NZ 6.1ords (Rev Computer Part No. 701879-00046 (NZ-2010)) or later: Performance with...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST01615WI-D. The requirements for environmental protection and the associated certified noise and/or emissions levels of the...</td>
<td>Applicable only to airplanes equipped with Honeywell SPZ-8000 Fight Control System (FCS) and EDZ-820 Electronic Flight Instrument System (EFIS) Displays with dual or triple NZ-2000 Flight Management System. Until such time that the TAWS Mode-S protection is available, the approval will be limited to...</td>
<td>30/03/2020</td>
<td>Active</td>
<td>10065316</td>
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<td>INTER-TEC AERO DESIGN AND CERTIFICATION LTD</td>
<td>Installation of pitch aircraft seating system models PF3300H and PF3300H5Rev. 2 Transfer of STC holder from INTER-TEC SERVICES Ltd to INTER-TEC AERO DESIGN AND CERTIFICATION LIMITED.</td>
<td>The Type and OSD Certification Bases for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this change/repair it must be determined whether the interconnectedness between the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/12/2021</td>
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<td>10065350</td>
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<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>CABIN MANAGEMENT SYSTEM &amp; WLAN IN EMB-135BJ [MSN 14501094]</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Amorworthiness Approval Limited to Airplane EMB-135BJ [MSN 14501094].</td>
<td>25/04/2018</td>
<td>Active</td>
<td>10065355</td>
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<tr>
<td>10065358</td>
<td>FUNT AERO, INC.</td>
<td>FAA STC SA2676WE</td>
<td>207/207A/T207/T207A</td>
<td>Installation of larger fuel containing wing tips.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC. There are requirements for environmental protection and the associated certified noise and/or emissions levels of the original product...</td>
<td>FAA issued Final Air Data List No. DL-FA200, Revision A, subsequent FAA sealed revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/04/2018</td>
<td>Active</td>
<td>10065358</td>
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<tr>
<td>10065359</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FALCON 900EX</td>
<td>Validation of FAA STC ST10862S/Airports and Interior Installations</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Data List Report No. 16067-16068, Revision B dated 18th – 25th July 2012. Aircraft Flight Manual Supplement Report No. 375159-1606716068 dated 17 July 2012. Supplemental Maintenance Manual Report SWM F9000EX-SN 250, Revision 01, dated 13 July – 2012 for later revisions of the above listed document(s).</td>
<td>This approval is limited to Falcon 900EX Serial Number 250. Prior to installation of this change it must be determined that the interrelationship between this change and any other previously installed change will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/04/2018</td>
<td>Active</td>
<td>10065359</td>
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<tr>
<td>10065360</td>
<td>AIRBUS DEFENCE and SPACE S.A.</td>
<td>A330-243</td>
<td>Installation of FSTA EF8 Fitting Device Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Airbus Defence and Space, doc. no. TEA-FL-DT-180008, Issue B, dated 23 March 2018 for later revisions of the above listed document(s) approved under the EASA system.</td>
<td>This STC can only be installed if one of the following EASASTC approval is previously installed: A330 FSTA STC1 10034690 or A330 FSTA STC2 10035940. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change...</td>
<td>25/04/2018</td>
<td>Active</td>
<td>10065360</td>
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<tr>
<td>10065366</td>
<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td>DHC-6 SERIES 400</td>
<td>Installation of AMT-1800 IG Antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List MD-2124-000000, Iss. 1- Master Drawing List DL-2124-000010, Iss. 1- Maintenance Manual Supplement MM-2124-000000, Iss. 1- for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>1/02/2021</td>
<td>Active</td>
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<td>10065371</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>REV. 1</td>
<td>P-68 &quot;VICTOR&quot;, P-68R &quot;VICTOR&quot;, P-68TC &quot;OBSERVER&quot;, P-68 &quot;OBSERVER 2&quot;, P-68 &quot;OBSERVER&quot;, P-68C, P-68C-TC</td>
<td>ISBAS enable / This design change approves the navigation using GPS and ISBAS with the installed Garmin GNS530/530X and P68/311.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>STC-114-0-AFM, Aircraft Flight Manual Supplement Revision 1, Transfer from ADOA to DOA.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/01/2021</td>
<td>Active</td>
<td>10065371</td>
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<tr>
<td>10065373</td>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>FAA STC ST0394AWT</td>
<td>Installation of TIMCO Aerosystems lavatories in an Airbus A330-200 Series aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>TIMCO Aerosystems Master Draw List (MDL) 10T1N05001, Revision R, dated 1 April 2012, or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/04/2018</td>
<td>Active</td>
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<td>STC Holder</td>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>FAA STC</td>
<td>ST03918AT</td>
<td>A330-343, A330-342, A330-341, A330-333, A330-332, A330-331, A330-330, A330-322, A330-321, A330-320, A330-319, A330-318, A330-317, A330-316, A330-315, A330-314, A330-313, A330-312, A330-311, A330-310</td>
<td>Installation of TIMCO Aeronautics lavatories in an Airbus A330-300 Series aircraft. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>TIMCO Aerospace Systems Master List (MDL) 10717NLS51, Revision R, dated 09 November 2012, or later FAA approved revisions. TIMCO Aeronautics Instructions for Continued Airworthiness 10717NRS51, Rev. A, dated 20 November 2012, or later FAA approved revisions.</td>
<td>Prior to installation of the design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce an adverse effect upon the airworthiness of the product.</td>
<td>27/04/2018</td>
<td>Active</td>
<td>10065534</td>
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<tr>
<td>STC Holder</td>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>FAA STC</td>
<td>ST04233AT</td>
<td>777-300ER</td>
<td>Interior reconfiguration installation for Boeing 777-300ER Series aircraft. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- TIMCO Aeronautics, LLC Master List 1470391P00, Revision J, dated 02 April 2018, or later FAA-approved revisions. - TIMCO Aeronautics (/&gt;-LLC Instructions for Continued Airworthiness 1470391P00, Revision D, dated 6 February 2018 and 1470391P05 Revision Dated 3 March 2018, or later FAA approved revisions.</td>
<td>Prior to installation of the design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce an adverse effect upon the airworthiness of the product.</td>
<td>27/04/2018</td>
<td>Active</td>
<td>10065537</td>
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<tr>
<td>STC Holder</td>
<td>B/E AEROSPACE, Inc.</td>
<td>FAA STC</td>
<td>ST04241AT</td>
<td>737-900-900ER 737-600-700-700CI-800</td>
<td>Installation of the Honeywell MK II Aircraft Communication Addressing and Reporting System (ACARS) Data Link Communication Function in the B737NG Series Aircraft. The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Drawing List (MDL) 2-153882-001, Rev. A, Dated 20 November 2017, or later FAA approved revision. 2. Airplane Flight Manual Supplement for the Honeywell MK II Aircraft Communication, Addressing (&lt;&gt;/&gt;) and Reporting System (ACARS) Data Link Communication Function (&gt;/&gt;), dated 2-153882-007, Rev. R.</td>
<td>Prior to installation of the design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce an adverse effect upon the airworthiness of the product.</td>
<td>30/04/2018</td>
<td>Active</td>
<td>10065534</td>
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<tr>
<td>STC Holder</td>
<td>AEROERVIS, S.R.O.</td>
<td></td>
<td></td>
<td>L-410 UVP-E, L-410 UVP-E9, L-410 UVP-E</td>
<td>Replacement of the original electric fuel booster pump by BP-DOA-06 pump. The Certification Basis for the original product remains applicable to this certificate/approval. Additional or alternative airworthiness requirements: Elect to Comply requirements CS-23 423.29, 23.301 (b), (c), (d), 23.305, 23.603, 23.607, 23.609, 23.611, 23.613 (b)(1), (b)(2), 23.863, 23.901 (b), 23.943, 23.951, 2.</td>
<td>- Master Document List doc. No. AE164000X, revision 0, dated 12.03.2018. L-410 UVP-E, -9 and -E20 Maintenance Manual (Do-L410-1222.2) Supplement doc. No. STC AE164-OTP1, issued November 2017. L-410 UVP-E, -9 and -E20 Wiring Manual (Do-L410-1242.2) Supplement doc. No. STC AE164-DAE1, issued 20 August 20...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will not introduce a adverse effect upon the airworthiness of the product.</td>
<td>02/05/2018</td>
<td>Active</td>
<td>10065403</td>
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<td>STC Holder</td>
<td>Rev Level</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>DOA AEGLOBAL LIMITED</td>
<td>REV. 1</td>
<td>777F</td>
<td>777-300</td>
<td>T-PED Tolerance CompatibilityRev 1. Transfer from AES Ltd to new STC holder AES Global Ltd.</td>
<td>- Master Data List, Ref AES-777-4653MDL. Issue 1  - Airplane Flight Manual Supplement, AES Supplement No. 77. Issue 1 for later revisions of the above listed document(s) approved under theEASA system.</td>
<td>- This modification provides the aircraft configuration management and existing requirements to facilitate demonstration of aircraft compatibility with PED emissions operating in the25-40GHz and 5G/HzFrequency range, within the flight deck, cabin and cargo area. Refer toMaster Data List AES-777-4653MDL...</td>
<td>28/04/2021</td>
<td>Active</td>
<td>10065407</td>
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<td>AERSALE, Inc.</td>
<td>REV. 3</td>
<td>FAA ST04000NY</td>
<td>737-600, 737-700, 737-800, 737-300, 737-400, 737-500</td>
<td>The Certification Basis for the original product as amended The following paragraph(s) at a later amendment: CS 25.1, 25.1309(a) 1.25.1520(a), 25.1629, 25.1581(a) at Amrd 20. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged.</td>
<td>1. Instructions for ContinuedAirworthiness, Doc Ref: B737-ICA-0019 RevB2 Flight Manual Supplement, Doc Ref: B737-0017Rev Cor later revisions of the above listed document(s) approved/ accepted onehalf of EASA in accordance with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>None</td>
<td>20/04/2020</td>
<td>Active</td>
<td>10065422</td>
<td></td>
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<tr>
<td>CRANFIELD AEROSPACE SOLUTIONS LIMIT</td>
<td></td>
<td>OSD for STC Number 10056170 Initial Issue of the MMEL Supplement for C525A</td>
<td>525A</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>CAS/MMEL1452, Master Minimum Equipment List Supplement, Installation of Tamrack ATLAS Winglets, Issue A, dated January 2018 for later revisions of the above listed documents approved by EASA.</td>
<td>Applicable to Cessna 525A aircraft fitted with EASA STC 10056170 issued Must be used in conjunction with MMEL S25.6ME1E/E from TCH Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and...</td>
<td>04/05/2018</td>
<td>Active</td>
<td>10065436</td>
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<tr>
<td>QUIET TECHNOLOGY AEROSPACE, Inc.</td>
<td></td>
<td>FAA STC ST04233AT</td>
<td>60</td>
<td>Replacement of the inner aluminum barrel of the engine inlet cowling, with an inlet barrel of carbon graphite sandwich construction, to provide avulsion for in-service corrosion.</td>
<td>QTA Assembly Procedure, Report QTA 6004, Revision A, dated 24 June 2015 QTA Master Drawing List, Report QTA 6001, RevA, dated 24 June 2015 Instructions for Continued Airworthiness, Report QTA-6005, Rev IR dated 31 December 2014 Structural Repair Manual Report, QTA-6006, Rev IR dated 16 January 2018 Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/05/2018</td>
<td>Active</td>
<td>10065432</td>
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<tr>
<td>JET AVIATION AG</td>
<td></td>
<td>VVIP Cabin Interior Completion</td>
<td>777-300</td>
<td>As per CRI A-01 The requirements for environmental protection and the associated certified noise and/or emissions level of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, Doc. No. JBSC-M-0056MDL-01 Issue 03-Certification Compliance Sheet. Doc. No. JBSC-M-0056CSS-01 Issue 03-Aircraft Flight Manual Supplement, Doc. No. JBSC-M-0056AFM-01 Issue 06-Airworthiness Limitation Section (ALS) of the ICA, Doc. No.:JBSC-M-0056AMT-01 Issue 01 or later revisions...</td>
<td>None</td>
<td>04/06/2018</td>
<td>Active</td>
<td>10065430</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>FAA STC/Doc No</td>
<td>Foreign/NAA Doc No</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>B&amp;G AEROSPACE, Inc.</td>
<td>REV. 1</td>
<td>FAA ST20967AT</td>
<td>737-800, 737-900, 737-700, 737-600, 737-500, 737-400, 737-300</td>
<td>Installation of an Avtech NA138-714BC Selective Calling (SELCAL) System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1, EMTEQ Engineering Master Drawing List (MDL) 2-27097-001 Rev. C dated 12 June 2006, or later FAA approved revision 2, FAA Approved Flight Manual Supplement (AFMS) for an EASA/FAA 1344/48C SELCAL System on Boeing 737 Series Aircraft, EMTEQ Document NO. 2-27097-838, Rev. 09, dated 30 March 2010, or...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2018</td>
<td>Active</td>
<td>10065438</td>
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<tr>
<td>365 AEROSPACE DOS LIMITED</td>
<td>REV. 1</td>
<td>PA-31T (CHEYENNE II)</td>
<td>PA-31T (CHEYENNE II)</td>
<td>Installation of Weather Modification Equipment</td>
<td>Rev. 1, Transfer of STC from STC Ref. PA-31T2 (CHEYENNE II) to 365 Aerospace Ltd. to 365 Aerospace DOS Ltd</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.</td>
<td>-365 Aerospace Ltd. Master Data List MDL-0234 Rev. C, issued 2 May 2018-365 Aerospace Ltd. Flight Manual Supplement, FMS-0234-001, Rev. A-365 Aerospace Ltd. Instructions for Continued Airworthiness CA-0234-001, Rev. A, issued 16 Feb. 2018 for earlier revisions of the above listed document(s) approved...</td>
<td>The installation of STC FAA SA1593GL is a preconidition for installation of the weather modification equipment. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/08/2019</td>
<td>Active</td>
<td>10065448</td>
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<tr>
<td>POLYTECH AERO SUPPORT INC.</td>
<td>REV. 1</td>
<td>TCCA SA17-52 ISSUE 1</td>
<td>A330-343</td>
<td>Installation of Mood Lighting System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>P.A.S. Modification Summary No. 17-002 R2, dated July 28, 2018 for earlier revisions of the above listed document(s) approved...</td>
<td>None</td>
<td>31/08/2018</td>
<td>Active</td>
<td>10065451</td>
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<tr>
<td>POLYTECH AERO SUPPORT INC.</td>
<td>REV. 1</td>
<td>TCCA SA17-53 ISSUE 1</td>
<td>A330-343, 330-342, 330-343</td>
<td>Installation of Mood Lighting System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>POLYTECH AERO SUPPORT INC. DOC. NO. MIS-17-003 REVISION 1 JUN 7 2018, or later revisions of the above listed document(s) approved...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/10/2018</td>
<td>Active</td>
<td>10065462</td>
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<tr>
<td>NORTH AMERICAN SURVEILLANCE SYSTEMS Inc.</td>
<td>FAA STC SA0031NY-D</td>
<td>208B</td>
<td>Installation of Camera Ports</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. Except where amended by additional or later amendments or supplements if indicated on FAA STC. Those requirements for environmental protection and the associated certificate/noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Delta Engineering Master Data List of 0160-08107-2321 Revision dated 5 September 2012 or later FAA approved revisions to Delta Engineering Master Data List previously listed. The instructions for Continued Airworthiness and airplane Flight Manual Supplement as listed on the Master Data List.</td>
<td>This STC can only be installed on aircraft that have the installed TKR-505 Protection System. The TKR-505 system is not approved by this STC prior to installation of this design change in addition to the STC installed on this design change. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change...</td>
<td>07/05/2018</td>
<td>Active</td>
<td>10065463</td>
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<tr>
<td>PALL AEROPRO CORPORATION</td>
<td>FAA ST04414AT</td>
<td>A321-271N/-272N, A321-251N/-252N/-253N, A320-231N/-232N/-233N, A320-211N/-212N/-213N/-214N, A319-151N/-152N/-153N/-154N, A318-151N/-152N/-153N/-154N</td>
<td>Installation of Pall Aerospace Corporation FTIS pre-filter kit part number A2690075STC.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Pall Aerospace Drawing No. A290957STC, Rev. A, dated June 1, 2017 FAA approved revisions.</td>
<td>The installation applies to the reissued aircraft models/daylight equipped with Parker IGS5 pallet part number 2070066-102 or 2070066-103 installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design...</td>
<td>07/05/2018</td>
<td>Active</td>
<td>10065460</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AIRBUS S.A.S.</td>
<td></td>
<td>10065491</td>
<td>A380-841, A380-842, A380-861</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. The Approval is limited to MSN 30867. If the holder agrees to permit another person to use this certificate holder, the holder must give the other person written evidence of that permission. The applicant's assessment showing that this modification does not impact the existing electrical wiring...</td>
<td>- MDS 5-05-4113 Issue 4, dated 27 April 2018. Prior to installation of this change/repair it must be determined that the installation or with any previously installed change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Data List, ref : MOL-5-05-4113 Issue 4, dated 27 April 2018.</td>
<td>14/06/2018</td>
<td>Active</td>
<td>10065491</td>
<td></td>
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<td>T.P.S. S.P.A.</td>
<td>10065501</td>
<td>10065502</td>
<td>AS 332 L1, AS 332 L</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Prior to installation of this change/repair it must be determined that the installation or with any previously installed change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>- Master Data List, ref : MOL-5-05-4113 Issue 4, dated 27 April 2018. Prior to installation of this change/repair it must be determined that the installation or with any previously installed change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Data List, ref : MOL-5-05-4113 Issue 4, dated 27 April 2018.</td>
<td>14/06/2018</td>
<td>Active</td>
<td>10065501</td>
<td></td>
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<tr>
<td>SABENA TECHNICS BOO</td>
<td>10065503</td>
<td>10065511</td>
<td>28 MARK 0100</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Prior to installation of this change/repair it must be determined that the installation or with any previously installed change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>- Master Data List, ref : MOL-5-05-4113 Issue 4, dated 27 April 2018. Prior to installation of this change/repair it must be determined that the installation or with any previously installed change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Data List, ref : MOL-5-05-4113 Issue 4, dated 27 April 2018.</td>
<td>14/06/2018</td>
<td>Active</td>
<td>10065502</td>
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<td>ALAMO ENGINEERING GmbH</td>
<td>10065511</td>
<td>10065512</td>
<td>BD-700-1A11, BD-700-1A10, BD-700-1A11, BD-700-1A10</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Prior to installation of this change/repair it must be determined that the installation or with any previously installed change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>- Master Data List, ref : MOL-5-05-4113 Issue 4, dated 27 April 2018. Prior to installation of this change/repair it must be determined that the installation or with any previously installed change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Data List, ref : MOL-5-05-4113 Issue 4, dated 27 April 2018.</td>
<td>14/06/2018</td>
<td>Active</td>
<td>10065511</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>10065512</td>
<td>10065513</td>
<td>FAA STC, SR06616RC</td>
<td>Installation of Floor Protectors</td>
<td>Master Data List, ref : MOL-5-05-4113 Issue 4, dated 27 April 2018. Prior to installation of this change/repair it must be determined that the installation or with any previously installed change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Data List, ref : MOL-5-05-4113 Issue 4, dated 27 April 2018.</td>
<td>14/06/2018</td>
<td>Active</td>
<td>10065512</td>
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<td>STC Holder</td>
<td>Rev.</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>Quiet Technology Aerospace</td>
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<td>ST4413AT</td>
<td>FALCON 200EX</td>
<td>Replacement or the inner aluminum barrel of the engine inlet cowl(+), without an inner barrel of carbon graphite sandwich construction(+), provide a solution for in-service corrosion.</td>
<td>1. QTA Note Civil A319-133 As of Falcon F2000, Drawing F200-NC-2000, Rev. NC, dated 29 September 2017, 2. Assemble Procedure, Report QTA-2006, Rev. IR, dated 16 August 2017, 3. Instructions for Continued Airworthiness, Report QTA-2006, Rev. IR, dated 13 September 2017, Structural Repair Manual, Report QTA-...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/05/2018</td>
<td>Active</td>
<td>10065514</td>
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<td>Quiet Technology Aerospace</td>
<td></td>
<td>ST4417AT</td>
<td>BD-100-1A10</td>
<td>Replacement of the inner aluminum barrel of the engine inlet cowl with Quiet Technology Aerospace inner barrel of carbon graphite sandwich construction.</td>
<td>1. Quiet Technology Aerospace document “Challenger 300/350 Series Inlet Cowl Inner Barrel Replacement Master Drawing List,” Report No. QTA 6301, Rev. IR, dated 1 December 2012, “Bombardier Challenger 350/350 Inlet Cowl Inner Barrel Replacement Assembly Procedure,” Report No. QTD 6305, Rev. IR, dated 1...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/05/2018</td>
<td>Active</td>
<td>10065515</td>
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<td>ICELANDAIR EHF</td>
<td>757-200 SERIES</td>
<td>757-200 SERIES</td>
<td>757-200 SERIES</td>
<td>Demonstrating the 120-minutes ETOPS capability per AMC-20-6 Revision 2 for RR211-63E4-powered Boeing 757-200 series aircraft in accordance with Icelandic CAA-757-0000-23, Initial Release dated 23 February 2018. With this STC, Icelandair demonstrated that with three reliable and independent electrical...</td>
<td>- Icelandair’s Change Repair Approval ref. CAA-757-0000-23, Initial Release dated 23 February 2018, Icelandair’s Continuing Airworthiness Instructions CAA-757-0000-23, Revision A, latest revision of the above listed document(s) approved under the EASA system.</td>
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<td>16/06/2018</td>
<td>Active</td>
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<tr>
<td>AAAC Aerospace Switzerland AG</td>
<td>REV. 1</td>
<td>A319-133</td>
<td>A319-133</td>
<td>Alternate Cabin reconfiguration with reduced passenger capacityRevision 1: The alternate configuration option to attirbutedenominate-installation of the seats 68B and 68K (scissored). The seatback permanently removed.</td>
<td>- The change is defined with Revision A of MDL-GF04-2520 amendment/Revision B of MDL-GF02-2520. The Aircraft shall be operated in accordance with Revision B of AMS-FMS-GF02-2520. The aircraft shall be operated in accordance with Revision B of AMS-FMS-GF02-2520. The aircraft shall...</td>
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<td>25/07/2019</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>Upgrade of Panasonic Avionics Corporation System Monitors</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST02660NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Data Provided Master Data List 0108-08700-2211 Revision B dated 17 April 2018 or later FAA approved.</td>
<td>This STC is for replacement of existing Panasonic RD- AA7204-01 monitor. Prior to installation of this design change, it must be determined that the interface between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft.</td>
<td>17/05/2018</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement Doc. No. STC-118-G-AMM Rev. 1, Date 5.5.2018. Prior to installation of this design change, it must be determined that the interface between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft.</td>
<td>Prior to installation of this design change, it must be determined that the interface between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft.</td>
<td>18/06/2019</td>
<td>Active</td>
<td>10065549</td>
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<tr>
<td>E.A.D AEROSPACE AIRWORTHINESS</td>
<td>IA-band SATCOM radome with provisions for LRU (antenna) installation on top of the aircraft fuselage.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List, EAD Aerospace document No. 01555-EAS-A-AMM-02, Revision 1, Date 03 May 2019. - Aircraft Flight Manual Supplement, EAD Aerospace document No. 01555-EAS-A-AMM-01, Revision A, dated 26 April 2018.</td>
<td>Prior to installation of this design change, it must be determined that the interface between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft.</td>
<td>17/06/2019</td>
<td>Active</td>
<td>10065550</td>
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<tr>
<td>PMV ENGINEERING</td>
<td>S-bandSatcom on A320 family. This modification installs mechanical/electrical provisions for the installation of a S-band Satcom system.</td>
<td>The Certification Basis for the original product as amended installs mechanical/electrical provisions for the installation of a S-band Satcom system.</td>
<td>- Master Document List, EAD Aerospace document No. 01555-EAS-A-AMM-02, Revision 4, dated 26 April 2018.</td>
<td>Prior to installation of this design change, it must be determined that the interface between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft.</td>
<td>12/11/2018</td>
<td>Active</td>
<td>10065567</td>
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<tr>
<td>BROMMA AIR MAINTENANCE AB</td>
<td>Bromma Air Maintenance AB Modification 16-10 for installation of MS7000 Airborne Maritime Special Mission System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Bromma Master Document List MDL-16-10 Revision A, dated 18 May 2018.</td>
<td>Prior to installation of this design change, it must be determined that the interface between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the aircraft.</td>
<td>07/01/2019</td>
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<tr>
<td>TELEDYNE BATTERY PRODUCTS</td>
<td>FAA STC 5006868</td>
<td>10065570 Installation of Teledyne Battery Products Value-Regulated Lead-Acid (VRLA) P/N 7638-44 main battery and P/N 1407-28 auxiliary battery.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>- Teledyne Battery Products Document No. TD72-1-01, Master Drawing List AB13946139.</td>
<td>Prior to installation of this design change it must be determined that the interchangeability between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/05/2018</td>
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<td>10065570</td>
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<tr>
<td>AIRBUS DEFENCE AND SPACE S.A.</td>
<td>REV. 1</td>
<td>10065571 OSD A330 MRTT FAF version Flight Crew Data Ref STC 10684192.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>Airbus Operational Suitability Data – Flight Crew, OSD-FC Final Report A330 MRTT FAF STC TAS-FF-RR-17-0004, version B, dated 25 January 2018.&lt;br&gt;None later revisions of the above listed document(s) approved under the EASA system.</td>
<td>28/06/2018</td>
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<td>10065571</td>
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<td>SABENA TECHNICS BOD</td>
<td>A310-304</td>
<td>10065575 A310 Cabin modification with new 55” screen. This modification consists in replacing an off-facing TV monitor in the lounge area by a new 55’ screen.</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS 25.758(b) or amendment(s).</td>
<td>Master Data List, ref. MDL-S-23-417 Revision A, dated 14 May 2018.&lt;br&gt;None later revisions of the above listed document(s) approved under the EASA system.</td>
<td>This modification needs the concurrent emboldenments of the following design changes as prerequisites: EASA STC 10043437 dated on 22 January 2013.&lt;br&gt;(Tarmac modification N° 687): PMV Minor Change, ref CAB-02440-MCA (supporting PMV Service Bulletin CAB-02440-SBB-501-R01). Prior to installation.</td>
<td>22/05/2018</td>
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<td>10065575</td>
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<td>DECA AVIATION ENGINEERING</td>
<td>SA 365 N, SA 365 N1</td>
<td>10065576 TCCA SH17.30, ISSUE 1 SA 365 N, SA 365 N1 EC 155 B, EC 155 B1 AS 365 N2, AS 365 N3 Nickel-Cadmium battery with Concorde P/N RG-386 Lead Acid battery.</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS 25.758(b) or amendment(s).</td>
<td>Concorde Battery Co. Modification Summary 5-4620, Revision B, TCCA Approved August 18, 2017.&lt;br&gt;None later revisions of the above listed document(s) approved under the EASA system in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined if the interchangeability between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/05/2018</td>
<td>Active</td>
<td>10065576</td>
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<td>SABENA TECHNICS BDC</td>
<td>A320-214</td>
<td>10065631 Rave IFE System Installation on A320</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s): F-01 (Pn 10035441), Securitisation (AirCraft systems and networks); the following paragraph(s) at a later amendment: Applicable regulatory climb.</td>
<td>- AFM Supplement 40166-01-A-4AFM-6380&lt;br&gt;None later revisions of the above listed document(s) approved under the EASA system, - Instructions for continued airworthiness 40166-01-A-ICA-R00,&lt;br&gt;Master Data List 40166-01-A-MLD-R00.</td>
<td>Prior to installation of this change/repair it must be determined if the interchangeability between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2018</td>
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<td>10065631</td>
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<td>JANZAIR CONSULTING SERVICES, INC.</td>
<td>FFA STC 737-200-300-400-500</td>
<td>10065632 Installation of a Dual Collins HF Communication System within Leading Edge Antenna into a Boeing 737-200/-300/-400/-500 aircraft.</td>
<td>Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC. Requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>Jaroszl Master Drawing No. 731-2311-101 Revision B, dated 15 July 2005, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined if the interchangeability between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2018</td>
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<td>EASA STC no.</td>
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<td>10065642</td>
<td>AIR ALLIANCE GmbH</td>
<td>35, 35A, 36, 36A</td>
<td></td>
<td>Replacement of GPS/WAAS Antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>According to Air Alliance GmbH Master Document LD2017 016 L335 Form 2, Rev.00, to be installed i.a.w. Air Alliance GmbH 2017 016 L335, 0500. Rev 00. To be maintained i.a.w.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product,...</td>
<td>25/05/2018</td>
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<tr>
<td>10065652</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES Inc.</td>
<td>REV. 1</td>
<td>777-300</td>
<td>Economy Class Reconfiguration and provisions for future approval optional stretch stretcher installation...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: 25.561, 25.791. 25.813(a)(b)(c)(e), 25.815, 25.1447 are at CS-25 Amendment 12 and 25.765is at CS-25 Amendment 13. The require...</td>
<td>- Certification Summary, Northwest Aeronautics Technologies, Inc. document/N0502-D06, Revision B, dated 20 May 2016; - Instructions for Continued Airworthiness, Northwest Aeronautics Technologies, Inc. document N0502-D70, Revision B, dated 7 May 2018 or later revisions of the above listed document(s) approved by the EASA system.</td>
<td>The following Approved B/E Aerospace Service Bulletins shall be concurrently installed on MSN aircraft 27504-27510: SB 25-30-1022, SB 25-30-1028, SB 25-30-1024, SB 25-30-1029 and SB25-30-1030. The following Approved ZPG Service Bulletins shall be concurrently installed on MSN aircraft 33702-33704, 342...</td>
<td>04/06/2018</td>
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<td>10065656</td>
<td>ELECTRONICS INTERNATIONAL, Inc.</td>
<td>FAA STC SAD20935E</td>
<td>SEE FAA AML</td>
<td>Installation of Electronics International Inc. Glass Panel EngineMonitor (with Flight Data) CGR-30P</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Installation Instructions, Document No. II 02191301, Revision B, dated 21 June 2013; or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. The STC is valid for aircraft types and...</td>
<td>29/05/2018</td>
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<tr>
<td>10065657</td>
<td>ELECTRONICS INTERNATIONAL, Inc.</td>
<td>FAA STC SAD24025E</td>
<td>SEE FAA AML</td>
<td>Installation of Electronics International Inc. Glass Panel EngineMonitor (with Flight Data) CGR-30C.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain...</td>
<td>1) Master drawing list as specified on AML, SA024025E, or later FAA approved revision.2) Electronics International, Inc. Instructions for Continued Airworthiness, Document No. 02131301, Revision B, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. The STC is valid for aircraft types and...</td>
<td>29/05/2018</td>
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<td>10065670</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td>REV. 1</td>
<td>767-200, 767-300, 767-400ER</td>
<td>Cabin monitor Installation on B767. EASA CS-25 Am0 20. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- AFM Supplement doc. NAV 0572 dated 01/02/2018; - MLO572 Rev. 5 (Master Document List)- ICA-NAV0572 Rev.2 (Instructions for Continued Airworthiness) or later revisions of the above listed document approved by EASA.</td>
<td>2. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. The STC is valid for aircraft types and...</td>
<td>22/06/2018</td>
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<td>10065672</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 1</td>
<td>FAA ST03199WY</td>
<td>Installation of ABC Completions Electronic Flight Bag Display Mounts andDAC International GDC-64- Configuration 1: EFB-Configuration 2: EBW with mount support InstallationRev 1. Update of Master Data List: EASA Certification basis corrected. Removal of dependency to prerequisite &quot;Quick...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 am0. 0: 25.853, 25.1583, 25.1585CS 25 am0. 5: 25.981. The requirements for environmental protection and the associated certified noise and/or emissions levels of the...</td>
<td>- Master Data List 0162-00039-2379, Rev. T, dated 02 February 2023; - Instructions for Continued Airworthiness No. 0162-00039-2379, Rev. B, dated 31 January 2023; - Airplane Flight Manual Supplement No. 0962-00038-2379, Rev. A, dated 01 February 2023 or later revisions of the above listed document(s) approved by the EASA.</td>
<td>1. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.2. This STC is for provisions of Electronic Flight B...</td>
<td>31/03/2023</td>
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<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>REV 1</td>
<td>TCGA-17-47 SS.1</td>
<td>MB-6-BK117 D-3, MB-6-BK117 D-2, MB-6-BK117 C-2, MB-6-BK117 C-2</td>
<td>Installation of Removable Pads for Hinged Door. Rev 1: Applicability extended to MB-6-BK117 D-3 helicopter model.</td>
<td>- Must be in accordance with Airbus Helicopter's Canada Limited Master/Drawing List MSL-17-001, Issue B.</td>
<td>Transport Canada approved 24th September 2021. Maintenance must be performed in accordance with Airbus Helicopter's Canada Limited Instructions for Continued Airworthiness (ICA-APCA-298, revision 1, ...</td>
<td>01/12/2022</td>
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<td>10225675</td>
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<tr>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>REV 5</td>
<td>CAA UK RESPONSIBILITY</td>
<td>EC135 T3</td>
<td>Powerline Mission Equipment and Load Carrying Components – ConfigurationAUK-135/3000 or AUK-135/3010 This STC includes, Anti-Spin Release System, External Cargo Information System, External Mirrors, Mission Radio and Load Carrying Components (LCC) including Workstation Baskets</td>
<td>Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-27 at Amendment 25-27 CS-27 CS27.605; 27.619; 27.625; 27.1585 / CS-27 at Amendment 3 for27.865 The requirements for enviro...</td>
<td>Initial Issue/Design Compliance Report(s): DCR/1353500-01 Issue A. Aircraft configuration for Power Line Operations: DCR/1353501-01 Issue A. Anti-Spin System: DCR/1353002-10 Issue A. Workstation Baskets &amp; Load Carrying Components: DCR/1353004-01 Issue B. Load Carrying Components: DCR/1353005-01 Issue...</td>
<td>08/12/2020</td>
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<td>SCANDINAVIAN AVONICS DESIGN ApS</td>
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<td>UB UPGRADING 1D 1290 The change consists of replacing the battery, in the existing (30 days) UBL, with a new Li-ion battery to support 90 days UBL operation.</td>
<td>Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition(s): See CRI-F-01 Issue 308 25 Amdt 2 for the affected requirements for environmental protection and the associated certificated noise and/or...</td>
<td>- Continued Airworthiness: Doc Ref: 21193-CAR later revisions of the above listed document(s) approved under 16EASA system.</td>
<td>04/06/2018</td>
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<td>10205686</td>
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<td>SABENA TECHNICS BOD</td>
<td>A330-341/242/243</td>
<td>A330-321/322/323</td>
<td>A330-301/302/303</td>
<td>A330-201/202/203/223/243</td>
<td>Landscape camera installation. This modification consists in mechanical and wiring routing provisions for the installation of a LATECOERE landscape camera, on lower fuselage/nest between FR(2) 2 and FR(3) 3, together with a plug (the plug is removed once the final installation of the camera).</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, ref: MDL-00-118 Issue dated 15 May 2018 Aircraft Flight Manual Supplement; ref: AFMS-F3059-00-011 Issue 1, dated 08 March 2018-Instruction for Continued Airworthiness, ref: ICA-F3059-00-009 Issue A, dated 23 April 2018 or later revisions of the above listed documents(s) approved u...</td>
<td>01/06/2018</td>
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<td>10205691</td>
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<tr>
<td>SABENA TECHNICS BOD</td>
<td>A330-323</td>
<td>A330-323</td>
<td>Aircraft configuration for Power Line Operations - A330-301/302/303/321/322/323/341</td>
<td>Landscape camera installation. This modification reconfigures the cabin layout of the MSN 758 by installing new passenger seats, UHCS, IFE and cabin features (curtains, rim, floor pathmarking, carpet...).</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, ref: MDL-S-00-4129 Issue A, dated 01 June 2018- Airworthiness Limitation Supplement; ref: ALS-F3058-00-141 Issue A, dated 30 May 2018-Instruction for Continued Airworthiness, ref: ICA-F3058-00-109 Issue A, dated 31 May 2018 or later revisions of the above listed documents(s) approved u...</td>
<td>The following modifications must be concurrently embodied with this modification: - EASA STC 1014567 Revision 2 (or later EASA approved revision); Floorpathmarking installation - EASA STC 1085691 (or later EASA approved revision): A330 landscape camera installation Prior to installation of this change...</td>
<td>01/06/2018</td>
<td>Active</td>
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An agency of the European Union
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<th>Foreign/NA</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<td>10065722</td>
<td>PEREGRINE LLC</td>
<td>FAA STC: SA00744DE</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of the Automatic Dependent Surveillance – Broadcast (ADS-B) Out Equipment using a Trim Avionics T73i Transponder and an approved position sensor.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC.</td>
<td>1. Permegene Master Drawing List, referenced on the attached AML No. SA00744DE. 2. Instructions for Continued Airworthiness document, as listed on the attached AML No. SA00744DE. 3. Airplane Flight Manual Supplement, as listed on the attached AML No. SA00744DE.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product. The STC is valid for those aircraft types and mo...</td>
<td>04/06/2018</td>
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<td>10065740</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC: SR06178RC</td>
<td>412EP</td>
<td>Installation of Enhanced Hover Hold.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>- Master Drawing List, Report No. AA-16038 Revision B, dated January 22, 2018 or later FAA approved revision. - Aeronautical Accessories Instructions for Continued Airworthiness Report No. AA-16045, No Revision, dated March 9, 2017, or later FAA accepted revision. - Aeronautical Accessories Rotorcraft F...</td>
<td>Eligible for installation on SN 37002 thru 37099 or aircraft modified with STC SR06178RC. Installation of Bell 407XP Pro Glass Cockpit and Pratt &amp; Whitney Canada PT67-T Twin Power Turboshaft Engines and associated systems. Prior to installation of this design change it must be determined that the...</td>
<td>05/06/2018</td>
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<td>10065747</td>
<td>ALPINE AEROTECH LP</td>
<td>TCCA STC: SH-18-1 ISSUE 1</td>
<td>505</td>
<td>Installation of Alpine Aerotech LP Bear Paw Kit in accordance with Master Data List AAL-390-024-001.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Bear Paw Kit Flight Manual Supplement AAL-390-025-401 Revision NHN&lt;+(-,+&lt;(+ or later revisions of the above listed document(s) approved on behalf of EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement. - Airworthiness Requirements Compliance Program Bell 505 Ba...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/06/2018</td>
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<td>10065748</td>
<td>STANDARD AERO</td>
<td>FAA STC: ST04236CH</td>
<td>GIV, GIV-SP</td>
<td>Installation of a Honeywell Jetwave satellite/associated Ka-Band Radome. This modification removes the existing Ku-Band radome and installs a new Ka-Band radome on the aircraft tail. It also installs the Honeywell Jetwave SATCOM System.</td>
<td>The Certification Basis for the original product is amended by the following alternative or additional airworthiness requirements: Equivalent Safety Finding(s): ESR B-101 Vibration/ buffeting/compliance criteria for large external antenna/ installation. The requirements for environmental protection and...</td>
<td>- Master Data List, ref. 1032265 Issue E, dated 06 JUN 2018. - Airplane Flight Manual Supplements, ref. 1032713 Rev A dated 22NOV2017. - Instructions for Continued Airworthiness, ref. 1032713 Rev A dated 22NOV2017 or later revisions of the above listed document(s) approved on behalf of EASA in accordance with the...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/06/2018</td>
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<tr>
<td>10065749</td>
<td>AVIONIK STRAUSBURG ENTWICKLINGS GmbH</td>
<td>FAA STC: SA00744DE</td>
<td>F172P</td>
<td>Installation of a special Antenna and Signal/Data processing Equipment for Mission Purposes.</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) of a later amendment: CS-23, Annex: 429, 414, 143, 149, 153, 171, 301, 303, 305, 307, 337, 561, 601, 603, 605, 607, 609, 611, 613, 625, 867, 13...</td>
<td>- Project Description ACSR-2017-085-PRD-04-00, Rev. 00. - Instructions for Installation ACSR-2017-085-IR-10-00, Rev. 00. - Instructions for Continued Airworthiness ACSR-2017-085-KCA-11-00, Rev 00</td>
<td>All provisions according to EASA STC 10065968 must have been already installed. This approval does not cover the installation of any mission/ research equipment in the 19&quot; rack other than specified Avionik Straubgldocument ACSR-2017-085-PRD-04-00. It does also not cover operation of any kind of PED P...</td>
<td>06/06/2018</td>
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<td>STC Holder</td>
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<td>STG AEROSPACE GERMANY GmbH</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>The Certification Basis for the original product as amended by the following additional alternative airworthiness requirements: Equivalent Safety Findings (s) See CR D-01. The following paragraph(s) at a later amendment: CS/55, Annex 20:25.1(a), 25.2(b), 25.27, 25.301, 25.303, 25.305(a)(6), 25,...</td>
<td>Master Data List Doc No SG-1012-MDL-01 Rev.</td>
<td>IR/AFM Supplement Doc No SG-1012-ICA-01 Rev.</td>
<td>For all aircraft with approved cabin crew Cabin Data Supplement Doc No SG-1012-ICA-01 and AFM Supplement Doc No SG-1012-FMS-01 Prior to in...</td>
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<td>BAE SYSTEMS AIRBUS GROUP</td>
<td>REV. 1</td>
<td>-5365851 Service Bulletin Issue 2-S36A Installation Design (Drawings) issue 2or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this design change it must be determined that the design change and any other previously installed design change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>01/02/2019</td>
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<td>HELIFAB Inc dba ARROW AVIATION CO. LLC</td>
<td>FAA STC SR06870AC</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the design change and any other previously installed design change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>07/06/2018</td>
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<td>JETECH, LLC</td>
<td>FAA STC ST0427LA</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the design change and any other previously installed design change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
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<td>SECUARPLANE TECHNOLOGIES, Inc</td>
<td>FAA STC SAU29666E</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the design change and any other previously installed design change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>11/06/2018</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>GVI (G650)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the design change and any other previously installed design change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
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<td>SABENA TECHNICS BGC</td>
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<td>A330-341, A330-342, A330-343, A330-321, A330-322, A330-323, A330-301, A330-302, A330-307, A330-243, A330-243F, A330-223, A330-222F, A330-201, A330-202, A330-203</td>
<td>Mechanical and electrical installations provisions for Thales EFISolutions in cockpit of A330. This STC deals with the installation in the cockpit on the LH and RH sides of windows of a mounting device and the electrical supply for Thales EFISolutions Rev 01. The option GROUP II (no equipped switch).</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s): 1) at a later amendment: Refer to sections 5 &amp; 6 of the installation plan (CS-25 Amendment 4). The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this STC.</td>
<td>Instructions for Continued Airworthiness, Aeroson document ref. 0866-D1-AFMS-R02A Appliance Flight Manual Supplement, Aeroson document ref. 0866-D1-AFMS-R03 Master Data List, Aeroson document ref. 0866-D1-AFMS-R02 Certification Plan, Aeroson document ref. 0866-D1-A-CP-R02 or later revisions.</td>
<td>Prior to installation of this change/repair it must be determined that the interference between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/11/2018</td>
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<td>DELTA ENGINEERING CORPORATION</td>
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<td>ST02646NY-D</td>
<td>Installation of a Panasonic Avionics Corporation System 3000i</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST02646NY. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this STC.</td>
<td>- Flight Manual Supplement FMS-150004-6 Revision C dated 16 May 2018; - Master Drawing List MLD-150004-3 Revision AD dated 16 May 2018; or later revisions of the above listed document(s) approved on behalf of EASA in accordance with the Technical Implementation Procedures of EASA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interference between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/06/2018</td>
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<td>AVIATION SPECIALTIES UNLIMITED, INC</td>
<td></td>
<td>AB139, AW139</td>
<td>Night Vision Imaging System (NVIS) Lighting Supplemental NVIS Lighting System manufactured and installed by Aviation Specialties Unlimited Inc for the helicopter models AB139 and AW139 in accordance with the listed Master Drawing List document MDL-150004-3 Revision AD or later FAA approved revisions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this STC.</td>
<td>- Flight Manual Supplement FMS-150004-6 Revision C dated 16 May 2018; - Master Drawing List MLD-150004-3 Revision AD dated 16 May 2018; or later revisions of the above listed document(s) approved on behalf of EASA in accordance with the Technical Implementation Procedures of EASA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interference between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>PATRIA AVIATION OY</td>
<td></td>
<td>AB412 EP</td>
<td>FMS and EHSI Installation Reason for Revision 1: limit the AFCS bank angle authority in GNSS-based NAV.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Flight Manual Supplement FMS-150004-6 Revision C dated 16 May 2018; - Master Drawing List MLD-150004-3 Revision AD dated 16 May 2018; or later revisions of the above listed document(s) approved on behalf of EASA in accordance with the Technical Implementation Procedures of EASA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interference between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/10/2019</td>
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<td>10040025</td>
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<tr>
<td>SPAES GmbH &amp; Co. KG</td>
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<td>204B, 205A-1</td>
<td>Installation of electrical Power / Control System Installation of an Power and Control Box for use of several external aids (e.g. towing line, bimini bucket) including change of pushbutton functions on the cyclic stick to operate the external aid.</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: FAR 29 including Amend. 4; FAR 29.21, FAR 29.25, FAR 29.27, FAR 29.29, FAR 29.251, FAR 29.301, FAR...</td>
<td>- FMS-SPAES-0072, Supplement for control box power and control system Revision 00 dated 3 May 2018; or later revisions of the above listed document(s) approved under the EASA system. - Installation Manual, Item: M-205-072-25-01, Revision 00 dated 3 May 2018 - Instructions for Continued Airworthiness.</td>
<td>Prior to installation of this change/repair it must be determined that the interference between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/06/2018</td>
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A agency of the European Union
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<tr>
<td>10065832</td>
<td>SPAES GmbH &amp; Co. KG</td>
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<td>AS 355 F2, AS 355 N, AS 355 NP, AS 355 E, AS 355 F1, AS 350 B1, AS 350 B2, AS 350 B3, AS 350 BA</td>
<td>Installation of automatic deployable ELT system series 503, manufactured by HRI-Smit consisting of emergency locator transmitter, beacon release unit, locator beacon, control unit, water switch and interface unit.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Part 145 PAL-355-008-25-01, rev. 00, issued on 9 April 2018, Aircraft Flight Manual Supplement FMS-SPAES-008 rev. 00, dated 13 June 2018; for later revisions of the above listed document(s) approved under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/06/2018</td>
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<td>10065834</td>
<td>BOMBARDIER Inc.</td>
<td></td>
<td>TCCA SA18-25 AT</td>
<td>BD-100-1A10</td>
<td>Installation, Lavatory Glass Mirror. EASA Validation of Transport Canada STC SA18-25 at issue 1.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): EASA Special Condition D-10553642-1 at Issue 1 (“Installation of Large Glass Items in the cabin”). The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Modification Data Summary LB4400 Revision NC, dated 29th May 2018; or later revisions of the above listed document(s) approved on behalf of EASA in accordance with the Technical Implementation Procedures of EU/Canada Bilateral Agreement.</td>
<td>Modification must be completed in accordance with Modification Data Summary LB4400 Revision NC, dated 29th May 2018; or later approved/ revisions. Maintenance must be in accordance with Instructions for Continued Airworthiness.</td>
<td>14/06/2018</td>
<td>Active</td>
<td>10065834</td>
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<td>10065841</td>
<td>QUIET TECHNOLOGY AEROSPACE, Inc.</td>
<td></td>
<td>FAA STC STD4261AT</td>
<td>GULFSTREAM 200/ GALAXY</td>
<td>Replacement of the inner aluminum barrel of the engine inlet cowl, within inlet barrier, for glass construction, to provide assistance for in-service corrosion.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>QTA Nose Cowl Assy - Gulfstream G200, Drawing G200-NC-6200, Rev. A, dated 14 March 2017, Assembly Procedure, Report QTA-6204, Rev. A, dated 23 January 2017, Instructions for Continued Airworthiness, Report QTA-6205, Rev. A, dated 23 February 2017, Structural Repair Manual, Report QTA-6206, Rev. A.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/06/2018</td>
<td>Active</td>
<td>10065841</td>
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<td>10065871</td>
<td>PATRIA AVIATION OY</td>
<td>REV. 1</td>
<td>ABA12EP ABA14 EP 412EP</td>
<td>E/OIR Camera &amp; Downlink / This STC consists in the installation of an eJetro Optical / Infrared (EO/IR) camera turret type model L3HarrisWescam MX-10 on the ABA12EP and Bell 412EP Helicopter together with avionics for Hand Controller for the turret - GPS Antenna for thetum - Operator’s console...</td>
<td>The Type and OSD Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>ABA12EP - Rotorcraft Flight Manual Supplement D-412-08-RFM-06 Issue 2, Revision 06 dated 8 June 2023 - Instructions for Continued Airworthiness D-412-08-ICA-01 Issue 2, Revision 05 dated 1 May 2018 and Bell 412EP - Rotorcraft Flight Manual Supplement D-412-08-RFM-14 Issue 1, Revision 06 dated 8 June 2023 - Instructions for Con...</td>
<td>- For all models: Available capacity for additional electrical load of 28VDC / 11 A. For model ABA14 EP: KI/PN 412-7718-29 FLIR Multispectral Turret System For model B412EP: Patria minor change D-412-08-DA-12 – Provision for EO/IR Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of this product.</td>
<td>11/07/2023</td>
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<td>1005890</td>
<td>GROB AIRCRAFT SE</td>
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<td>Flight Manual Supplement Doc. No. G.C208-01-1-ST1, Issue. For later revisions of the above listed document(s) approved under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/06/2018</td>
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<td>HEU AUSTRIA GmbH</td>
<td>REV. 1</td>
<td>AS 350 B3, EC 130 T2</td>
<td>Digital Backup Attitude Indicator and Digital Clock / USB Charger Reason for Revision 1: Effectively extended to EC 130 T2 and addition of second Digital Clock / USB Charger/PN.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Drawing List: Doc. No. HAU-D-042-001, Rev. 1; Rotorcraft Flight Manual Supplement, Doc. No. HAU-D-042-005, Rev. 1; Installation Manual, Doc. No. HAU-D-042-004, Rev. 1; Instructions for Continued Airworthiness, Doc. No. HAU-D-042-006, Rev. 1 or later revisions of the above listed docs...</td>
<td>This change is only applicable to AS350 B3, equipped with the GarminG500H cockpit, Mod No. 074723 and EC130 T2 helicopters, equipped with Garmin G505H cockpit, Mod. No. 074629 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and...</td>
<td>21/02/2019</td>
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<td>GFD GmbH</td>
<td>REV. 2</td>
<td>36A 35A, 36A 35A 35A</td>
<td>GPS Antenna Installation ProvisionsRev. 2; Transfer to GFD GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>According to AEC Master Data List No. 5565-048; Rev. A, To be installed i.e. AEC drawing List 5565-082, Rev. A, To be maintained i.e. AEC ICA document no. 5565-085, Rev. IR, or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
<td>Active</td>
<td>10059090</td>
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<tr>
<td>E.A.D AEROSPACE AIRWORTHINESS</td>
<td>REV. 1</td>
<td>A330-341, A330-342, A330-343 A330-321, A330-322, A330-323 A330-301, A330-302, A330-303 A330-223, A330-243 A330-201, A330-202, A330-203</td>
<td>Transmitting Portable Electronic Devices (T-PED) tolerance demonstration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Drawing List, EAD Aerospace document No. 01553-EAS-A-MDL, Revision A, dated 12 June 2018; Instructions for Continued Airworthiness, EAD Aerospace document No.01553-EAS-A-ICA-01, Revision A, dated 12 June 2018; Aircraft Flight Manual Supplement, Doc. No. 01553-EAS-A-AATM-01...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/06/2018</td>
<td>Active</td>
<td>10059026</td>
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<td>SIRIUM AEROTECH S.L.</td>
<td>REV. 1</td>
<td>747-400</td>
<td>SIRIUM AEROTECH S.L. Modification SA18-035-MLD revision (01). PED Tolerance/Compliance demonstration in accordance with SIRIUM/AEROTECH S.L. Master/Data List SA18-035-MLD revision (01). Thedesign change contains no physical change to the aeroplane(s) butconsists in showing that the aeroplanes are found toler...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- SIRIUM AEROTECH S.L Master Data List - PED Tolerance Compliance – SA18-035-MLD, Revision 1 – 8 May 2018 for later revisions of the above listed document(s) approved under the EASA system.</td>
<td>- This STC is limited to Boeing 747-400 series aeroplanes MEIH (Manufacturer Serial Number): 25/103, 26/450, 27/178, 28/427, 28/435 and 29/610. - Wireless communication must be limited to a maximum of 100m Equivalent Isotropic Radiated Power (EIRP). (for instance Wifi emissions limited to the 2.4 GHz ISM band...</td>
<td>14/08/2018</td>
<td>Active</td>
<td>10059209</td>
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<tr>
<td>BIGORRE AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAAST0462CH</td>
<td>FALCON 2000, FALCON 2000EX Installation of Cabin LED Lighting System Rev. 1; Transfer from GDS AERO LLC to BIGORRE AEROSPACE CORPORATION.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Instructions for Continued Airworthiness ref. doc. GDSLED-ICA, Rev. A; dated 2nd April 2015 - Electrical Load Analysis ref. doc. GDSLED-EIA, Rev. A, dated 22nd May 2015; Master Data List ref. doc. GDSLED-MDL, Rev. C, dated 12th January 2016 for later revisions of the above listed document(s) approved under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/06/2021</td>
<td>Active</td>
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<tr>
<td>ATI ENGINEERING SERVICES, LLC</td>
<td>FAA STC SUA03830</td>
<td>300/ 300LW 3 B300/ B300C Installation of LED Position Lights.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation / Manufacturing Drawing List DL-141211. Revision -. dated 25 September 2015, or later FAA approved revision/Instructions for Continued Airworthiness (ICA) ICA-141211, Revision dated 15 October 2015, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2018</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>STC Holder</td>
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<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td></td>
<td>SEE AML DOC. 2112821-EO-APP-2</td>
<td>Upgrade Underwater Locator Beacon (ULB) to upgrade to DK120/90</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s) on:</td>
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<td>22/06/2018</td>
<td>Active</td>
<td>10065950</td>
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<td>ALOFT AEROARCHITECT S</td>
<td>FAA STC ST03747NY</td>
<td>A320-201-202-203-223-245</td>
<td>Viasat KA SATCOM Antenna System Installation</td>
<td></td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated airworthiness levels of the original product remain applicable to this certificate/approval.</td>
<td></td>
<td></td>
<td>09/11/2018</td>
<td>Active</td>
<td>10065951</td>
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<td>BOGGI AERONAUTICS S.R.L.</td>
<td>REV. 1</td>
<td>A321-211, A321-231 A319-112, A320-214</td>
<td>Project 169P001-41 Equipment Bells for A320-214 MSNs equipped with ADF2.</td>
<td></td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated airworthiness levels of the original product remain applicable to this certificate/approval.</td>
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<td>20/09/2018</td>
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<td>BOGGI AERONAUTICS S.R.L.</td>
<td>A9139</td>
<td>AB139</td>
<td>A9139 EMS Interiors</td>
<td></td>
<td>Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated airworthiness levels of the original product remain applicable to this certificate/approval.</td>
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<td>26/06/2018</td>
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<td>10065955</td>
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<td>SABENA TECHNICS B&amp;G</td>
<td>REV. 2</td>
<td>A330-203, A330-223, A330-243</td>
<td>Seat Rails provisions for 9-abreast seat Rev. 01. This revision is to add two new models A330-203 and A330-223/23 to the original STC approval. Rev. 02: Editorial corrections.</td>
<td></td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraphs(s) at a later amendment: Applicable regulatory CS 25 Amendment 4 (refer to associated Certification Pave1212-C3-AC-P-R66, sections 5 and 6 for).</td>
<td>ICA - 40112-03-A-ICA- R62- Master Data List 40112-03-A-MDL-RG for later revisions of the above listed document(s) approved/accredited under the EASA system.</td>
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<td>08/11/2018</td>
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<td>Limitations/Conditions</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>A350-941, A350-1041</td>
<td>Dual Medical stretcher installation on A350.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: 25.3, 25.30, 25.303, 25.306, 25.307, 25.321, 25.331, 25.337, 25.341, 25.561, 25.613, 25.625, 25.785, 25.786, 25.789, 25.80...</td>
<td>- Master Data List AE17185MDL Rev. 1 dated 27/06/2018; FAA Supplement AE17185AFMS Rev. 0 dated 27/06/2018; Instructions for Continued Airworthiness AE17185CA Rev. 0 dated 27/06/2018 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>- Per AFM Supplement AE17185AFMS; - Per Instructions for Continued Airworthiness AE17185CA; - Pre-requisite: Aerotec Change AE16077MAS: This STC constitutes an approval for installation only for aircraft serial numbers as stated in drawing STR-16077-001-000. For additional aircraft serial numbers filed...</td>
<td>28/06/2018</td>
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<td>10066006</td>
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<td>AVIO DESIGN GROUP INC dba AVIO DG</td>
<td>TCCA SH18-22</td>
<td>Installation of Enhanced Vision System (EVS) Max-VIZ Model 1500 and 2300.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: FAR 27 Amdt. 27-40 for the affected requirements and FAR 27 Amdt. 27-42 for 27.1371 requirement. The requirements for envoir...</td>
<td>- Flight Manual Supplement 32.1404 Revision 2, dated October 10, 2012 for MAX-VIZ 1500 Enhanced Vision System (EVS)- Flight Manual Supplement 1740-FM-001 Rev. --, dated February 15, 2018 for MAX-VIZ 2300 Enhanced Vision System (EVS)- Flight Manual Supplement 1740-FM-002 Rev. --, dated February 15, 2018.</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/06/2018</td>
<td>Active</td>
<td>10066008</td>
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<tr>
<td>NORTWEST AEROSPACE TECHNOLOGIES NC.</td>
<td>A330-202</td>
<td>Interior Reconfiguration QANTAS A330-200 Rev. 1: Crew Rest Compartment design update.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI D-4304-002 Crew Rest Compartment. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product...</td>
<td>- Certification Summary: NO610-D90, Revision - , dated 27 June 2018 - Certification Summary: NO610-D91, Revision - , dated 7 September 2019 - Instructions for Continued Airworthiness: NO610-D70, Revision E, dated 22 August 2019 for later revisions of the above listed document(s) approved/accepted under the EA.</td>
<td>This STC shall be installed in conjunction with EASA STC 1005219. Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/09/2019</td>
<td>Active</td>
<td>10066011</td>
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<tr>
<td>SECURAPLANE TECHNOLOGIES, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST02580SE ERU 190-100 STD ERU 190-100 LL ERU 190-100 IGN ERU 190-100 EC2 Installation of two Securaplane Model 9750W0360-12-24 VDC28 Ah sealed lead acid batteries</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List AM0311-DL02-00 Revision A, dated 18 May 2018 or later FAA-accepted revision. The allowed aircraft must be maintained in accordance with the instructions for Continued Airworthiness (CA) document AM0311-IC07-00, Revision A, dated 24 April 2018 or later FAA-accepted revision.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/07/2018</td>
<td>Active</td>
<td>10066021</td>
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<tr>
<td>SABENA TECHNICS BOD</td>
<td>A330-223</td>
<td>A330-223 cabin and systems modification, and Ultra High Comfort# Stainless Seat installation. This modification reconfigures the cabin layout of the MSN 253 by installing new passenger seats, IUCAS, IFE and cabin features (curtain, ring, floor path marking, carpet...).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, ref.: MDL-5-00-4136 Issue A, dated 26 June 2018 - Airworthiness Limitation Supplement, ref.: ALS- F3599-00-125 Issue B, dated 01 June 2018 - Instructions for Continued Airworthiness, ref.: ICA- F3599-00-109 Issue E, dated 30 May 2018 - Aircraft Flight Manual Supplement, ref.: AFMS-F30...</td>
<td>The following modifications must be concurrently embodied with the modification: - EASA STC 10014507 Revision 2 (or later FAA approved revision): Floorpath marking installation - EASA STC 10065691 (or later FAA approved revision): A330 hashtag摄像头 installation Prior to installation of this c...</td>
<td>29/06/2018</td>
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<td>STC Holder</td>
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<td>Certification Basis</td>
<td>Description</td>
<td>Associated Technical Documentation</td>
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<td>10066026 SABENA TECHNICS BGC</td>
<td>Active</td>
<td>SABENA</td>
<td>29/06/2018</td>
<td>Pre-installation</td>
<td>CABIN RECONFIGURATION WITH ISPSS INSTALLATION</td>
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<td>10066031 ONBOARD SYSTEMS INTERNATIONAL</td>
<td>Active</td>
<td>FAA STC SH4904MM</td>
<td>04/07/2018</td>
<td>Pre-installation</td>
<td>Installation of Onboard Systems International Load Weigh Kits</td>
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<td>10066032 STC TWENTY ONE LIMITED</td>
<td>Active</td>
<td>CAA UK RESPONSIBILITIES</td>
<td>18/07/2018</td>
<td>Pre-installation</td>
<td>Introduction of structural provisions for GNSS antennas.REV. 2: editorial correction</td>
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<td>10066036 AEROTEC &amp; CONCEPT</td>
<td>Active</td>
<td>EASA STC</td>
<td>02/07/2018</td>
<td>Pre-installation</td>
<td>ASPEN EF/1000 primary flight display</td>
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**Notes:**
- CERTIFICATION BASIS:
  - For STC TWENTY ONE LIMITED:
    - The Certification Basis (CB) for the original product remains applicable to this certificate/approval.
    - The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.
  - For AEROTEC & CONCEPT:
    - The Certification Basis for the original product remains applicable to this certificate/approval.
    - The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.

- LIMITATIONS/CONDITIONS:
  - Prior to installation of any changes/replacements, it must be determined if the related installation will introduce any adverse effects upon the airworthiness of the product.
  - To remain compliant with FAA AD 2017-22-04 perform inspections per Service Bulletin, this design change does not introduce any adverse effects upon the airworthiness of the product.

- REMARKS:
  - STC REF. 40090-01-A-CP-R01 or later revisions of the above listed document(s) approved/accepted under the EASA.
  - Master Drawing List (MDL) No. 155-002-00, Revision 19, dated 9 October 2017, or later Federal Aviation Administration (FAA) approved revision.
  - To remain compliant with FAA AD 2017-22-04 perform inspections per Service Bulletin.
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<th>STC Holder</th>
<th>Model</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>SIA SOLUTIONS FOR AVIATION, S.L.</td>
<td>BNGA-21</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: paragraph(s) at a later amendment: CS 23 Amdt. 24 paragraph(s): 23.1, 23.21, 23.26, 23.29, 23.45, 23.49, 23.51, 23.53, 23.59, 23.63, 23.66, 23.73, 23.75, 23.141.2.</td>
<td>Flight Manual Supplement BN05S-HD001, Issue 1, dated 27/06/17 or later revisions of the above listed document(s) approved under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/07/2018</td>
<td>Active</td>
<td>10066037</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>A340-313</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Société Air France Change Approval Sheet CAS-A340-17-013 Revision dated 15 June 2018; Société Air France Cabin Crew Operating Manual Supplement COCOMS-A340-17-001 Revision dated 13 June 2019; New later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/07/2018</td>
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<td>10066039</td>
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<td>BOGGI AERONAUTICS S.R.L.</td>
<td>AS 365 N3</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Doc. No. PAL-CPR-1303 “Compliance Report”, Rev A dated June 27, 2018; Doc. No. PAL-1303 “Master Document Lot/Doc. No. PAL-PMS-AS365-112 “Flight Manual Supplement Trakkabeam® A800 Searchlight System”, Revision A; Doc. No. PAL-ICA-1303-001 “Instruction for Continued Airworthiness Trakkabeam® A800 Sea.”</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/03/2024</td>
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<td>BOGGI AERONAUTICS S.R.L.</td>
<td>AS 365 N2</td>
<td>Installation of Trakkabeam A800 High Intensity Searchlight System Rev. 1.1, STC transferred to Boggi Aeronautica S.r.l.</td>
<td>Master Document Lot/Doc. No. PAL-PMS-AS365-112 “Flight Manual Supplement Trakkabeam® A800 Searchlight System”, Revision A; Doc. No. PAL-ICA-1303-001 “Instruction for Continued Airworthiness Trakkabeam® A800 Sea.”</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/03/2024</td>
<td>Active</td>
<td>10066060</td>
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<td>DECA AVIATION ENGINEERING Ltd/b/a PAL AEROSPACE</td>
<td>TCCA O-LSIA18-009/D</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List 139P01410-MDL-03; Flight Manual Supplement 139P01410-FMS-02; or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>It is prohibited to operate the helicopter during any phase of flight and ground prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design...</td>
<td>18/01/2019</td>
<td>Active</td>
<td>10066054</td>
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<td>BOGGI AERONAUTICS S.R.L.</td>
<td>AS 365 N3</td>
<td>Installation of Trakkabeam A800 High Intensity Searchlight System Rev. 1.1, STC transferred to Boggi Aeronautica S.r.l.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design...</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>SWS CERTIFICATION SERVICES LTD.</td>
<td>777-300ER</td>
<td>SkyCouch Installation</td>
<td>The Certification Basis for the original product as amended by supplementary or alternative airworthiness requirements CS-25.120(a)(2)(ii) at Amrd 0. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable.</td>
<td>Master Configuration Scheme, Ref. 3168- MCS-01, Issue 2, or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>SkyCouch can only be used as a sleeping or accommodation/entertainment during-flight conditions, as detailed in SWS Certification Services Ltd Design Report 3168-DR-01. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/07/2018</td>
<td>Active</td>
</tr>
<tr>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>737-800-800-900ER</td>
<td>737 Family ADS-B Out - Honeywell</td>
<td>The Certification Basis for the original product as amended by supplementary or alternative airworthiness requirements CS-ACNS Initial Issue. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable.</td>
<td>Master document list MLD/1014-001 Issue 25. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/07/2018</td>
<td>Active</td>
<td>10066006</td>
</tr>
<tr>
<td>DELTA Q DESIGN Inc.</td>
<td>REV. 1 FAA STC</td>
<td>737-800-900ER</td>
<td>Installation of Securaplane and Otonomy cameras installation on Securaplane and Otonomy Cameras (structural provisions) on the tail and belly of the fuselage on Boeing 737 series aircraft.</td>
<td>- Master Data List, DDS- SOCM01-MD-01, rev. IR, dated 17 December 2015 - Instructions for Continued Airworthiness, DDS- SOCM01-SC-01, rev. IR, dated 15 December 2015 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Functional installation of cameras and associated equipment is not approved as part of this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/07/2018</td>
<td>Active</td>
</tr>
<tr>
<td>OUTERLINK CORPORATION</td>
<td>FAA STC</td>
<td>EC 130 B4</td>
<td>The STC installs a SATCOM and Flight Data Monitoring System</td>
<td>Model AS 350 B3 Outerlink Corporation Master Drawing List document number/MDL350-SAT-100, revision C, dated 10 February 2017, or later FAA approved revision 2.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/07/2018</td>
<td>Active</td>
</tr>
<tr>
<td>JET AVIATION AG</td>
<td>REV. 1</td>
<td>747-400F, 747-747SP</td>
<td>Rev. 0 The ADS-B Out installation consists of - replacement/upgrade of dual ATC SSR transponders by ADS-B mode S/RTCA DO260B - replacement of dual GNSS receivers by SBAS receivers or reuse of existing SA-Aware GPS receivers - replacement of dual GPS antennas by SBAS models when GPS receivers are not used</td>
<td>- Aircraft Flight Manual Supplement, JBSL-M-1096 AFM-01, Issue 3, Instructions for Continued Airworthiness, JBSL-M-098BDA-01, Issue 3. Engineering Order. JBSL-M-0108EEO-01, Issue 3.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/04/2019</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
<td>Model</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td></td>
<td>FAA STC 010436NY</td>
<td>767-8, 767-0</td>
<td>Installation of a Panasonic Avionics Corporation Global CommunicationSuite (GCS)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Data Engineering Master Data List 0165-10376-2433 Revision dated 08/05/2018 or later FAA approved revisions to Data Engineering Master Data List previously listed. The instructions for Continued Airworthiness as listed on the Master Data List is required with this installation.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>328 SUPPORT SERVICES GmbH</td>
<td>REV. 1</td>
<td>MYSTERE FALCON 50 (MF50)</td>
<td>Aviator 300 SBB SATCOM with WLAN.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>328 Support Services GmbH Airplane Flight Manual Supplement No. AFMS0200C 23/12/2001 Rev. B(&lt;&gt;)&lt;&gt;)the later revisions of the above listed documents approved by EASA-328 Support Services GmbH MASTER DOCUMENT LIST No. DL.1252001 Rev. B.</td>
<td>Restricted to ASN 345.</td>
<td>11/07/2018</td>
</tr>
<tr>
<td>JET AVIATION AG</td>
<td>REV. 1</td>
<td>SEE EASA-APPROVED MODEL LIST</td>
<td>WIRELESS LAN (WLAN) Activation Rev. 0 The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: Rev. 0 and 1: CS25 Amend 21, 25.789, 25.831, 25.859, 25.1301, 25.1309, 25.131, 25.1383, 25.1387, 25.1390, 25.1413, 25.1529, 2...</td>
<td>For Ref. 0 and 1: This change is applicable to airplanes previously fitted with properly certified Wireless Access Points(s) capable to transmit only on IEEE802.11a/b/g/n/ ac protocols, or a subset of these protocols with not more than 1000 mW transmitted power capacity. The WAP power supply must be sup...</td>
<td>03/11/2020</td>
<td>Active</td>
<td>10066134</td>
</tr>
<tr>
<td>JET AVIATION AG</td>
<td>REV. 1</td>
<td>747-400</td>
<td>747-300</td>
<td>WIRELESS LAN (WLAN) Activation1. Certification of Compatibility with Wireless Data Technologies shall be in accordance with JET Aviation AG Master Data List JBSL-M-01097MDL-01 Issue2, dated Aug 21, 2018 or later EASA approved revision. This STC demonstrates only to the immunity to IEEE 802.11 a/b/g/n/ ac pa...</td>
<td>NEW-PAGE The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25 Amend 21, CS 25.789, 25.831, 25.859, 25.1301, 25.1309, 25.1383, 25.1387, 25.1390, 25.1413, 25.1529, 25.1681, 25.7707 The requirements for environmental...</td>
<td>AFMS JBSL-M-01123AFM-01 Issue 1 &quot;WIRELESS LAN (WLAN) ACTIVATION&quot; Rev. 1: AFMS JBSL-M-01123AFM-01 Issue 2 &quot;FOR A/C Dassault F900EX/F2000/F2000EX/FXTS Series; WIRELESS LAN (WLAN) ACTIVATION&quot; AFMS JBSL-M-01123AFM-02 Issue 1 &quot;FOR A/C Bombardier BD-100-1A10/B-700-1A10/1A10/1A10-1A2-12 Series; WIRELESS LA. 1. This STC does not constitute installation approval of a wireless radio frequency system using wireless data technologies. It only provides data that may partially support their installations or activations with a separate approval. These data are provided in the Engineering Order JBSL-M-01097MDL-01...</td>
<td>24/08/2018</td>
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<tr>
<td>PRECISE FLIGHT, Inc.</td>
<td></td>
<td>FAA STC 0105905E</td>
<td>407</td>
<td>Installation of the Precise Flight Pulselite Controls System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The system is to be manufactured in accordance with Precise Flight Engineering Drawing lists 002P0000&gt;(&lt;&gt;)&lt;&gt;)Revision J, 01SP0000, Revision H, and/or FAA approved revisions. It is to be installed in accordance with appropriate Precise Flight Drawing List 141P0000 Revision B, or l...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>STC Holder</td>
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<td>STC Holder</td>
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<td></td>
<td>MYSTÈRE FALCON 900</td>
<td>Jet Aviation AG JBSL F900 Medevac Units Installation of Medevac units, medical equipment and medical cabinet.</td>
<td></td>
<td>- Master Data List JBSL-M-0098FDL-01 Issue 1 dated 9 July 2016 - AFM Supplement No. JBSL-M-0098FDL-01 Issue 2 dated 9 July 2016 - Instructions for Continued Airworthiness JBSL-M-0098FDL-01 Issue 2 dated 27 April 2018 or later revisions of the above listed document(s) approved by the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>FOKKER SERVICES B.V.</td>
<td>REV. 1</td>
<td>737-900, 737-900ER</td>
<td>10066148</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval without any influence.</td>
<td></td>
<td>- Compliance Record CR0043 Issue 1 dated 27 June 2018 - Aircraft Flight Manual Supplement B737-ADSB-AFM-S-001 Issue 2 dated 26 April 2018 - FOD Supplement B737-ADSB-FCD-S-001 Issue 1 dated 06 March 2018 - MMEL Supplement B737-ADSB-MMEL-S-001 Issue 1 dated 12 March 2019 For Revision 1 of STC-Cer...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>Q.C.M. DESIGN GmbH</td>
<td></td>
<td>BD-700-1A10</td>
<td>10066150</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval without any influence.</td>
<td></td>
<td>- Certification Report BB-D7-QCMS5-016, Rev. 1 Instructions for Continued Airworthiness, BB-D7-QCMS5-016-M1-ICA. Initial Issue or later revisions of the above listed document(s) approved/ acceptable to the EASA system.</td>
<td>Effectivity: S/N 9043 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>ICELANDAIR EHF</td>
<td></td>
<td>757-200, 757-300</td>
<td>10066156</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval without any influence.</td>
<td></td>
<td>- Icelandair Instructions for Continuing Airworthiness Document RA-757-4620-009 Initial Release, dated 05 July 2018 - Aircraft Flight Manual Supplement N-CA-757-4620-009 Issue 1 dated 27 April 2018 or later revisions of the above listed document(s) approved/ acceptable to the EASA system.</td>
<td>Prior to or concurrent modification (a): The EFB System is to be installed in accordance with ScandinavianAircraft Design (SAIAESA) STC 10059177. The GEE Wifi System (previously known asRow44) is to be installed per EASA STC 10044038 Icelandair DOA approved Minor changes EO-757-4620-05 andEO-757-4620-007...</td>
</tr>
<tr>
<td>N2 DEVELOPMENT, Inc.</td>
<td></td>
<td>A321-272NX</td>
<td>10066160</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval without any influence.</td>
<td></td>
<td>- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to or concurrent modification (a): The EFB System is to be installed in accordance with ScandinavianAircraft Design (SAIAESA) STC 10059177. The GEE Wifi System (previously known asRow44) is to be installed per EASA STC 10044038 Icelandair DOA approved Minor changes EO-757-4620-05 andEO-757-4620-007...</td>
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<td>STC Holder</td>
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<td>OUTERLINK CORPORATION</td>
<td>FAA STD SR003680</td>
<td>407 206L-4 206L-3 206L-1</td>
<td>Helicopters modified with an installation of an Outerlink SATCOM Communications and Flight Data Monitoring System Kit B206-ST-100 or KitB407-ST-100.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SR003680. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. Outerlink Corporation’s Master Drawing Lists (MDL); Document number MDL306- SAT-100, revision C, dated 9 March 2016 or MDL407- SAT-100; revision C dated 15 March 2016, or later FAA approved revision. Outerlink Corporation’s Rotorcraft Flight Manual Supplements (RFMS) Document number FMSB06061018R.</td>
<td>This system does not meet the operational rule requirement of a voice/flight data recorder system. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/07/2018</td>
</tr>
<tr>
<td>2 EXCEL DESIGN LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>200</td>
<td>Approval for the relocation of ADF Antenna on King Air 200 in accordance with 2 Excel Design Limited Modification 2XL051.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with 2 Excel Design Master Data List MDL3051 LFT18, Issue 2, dated, 06.07.2018. There is no Flight Manual Supplement associated to this change, Instructions for Continued Airworthiness (ICA) in accordance with 2Excel Design Document ref 2XL051 LFT731(I), Issue 1or later...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/07/2018</td>
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<tr>
<td>FTI ENGINEERING NETWORK Gmbh</td>
<td>B. B200GT, B300, B300C B. B200, B200C, B200CST A. G007T</td>
<td>Installation of ADS-B Out for King Air 900/500/300 and Premier 1/1A aircraft in accordance with FTI Engineering Network GmbH modifications ref 18-1632. C5-ACNS Issue 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with FTI Engineering Network GmbH modification ref 18-1632 Master Data List F18- EO18-1632, Issue 01 dated 09.07.2018 and Installation Instructions ref F23-001- EO18-1632, Issue 2, dated. 06.07.2018. Operation in accordance with FTI Engineering Network Contact Aircraft Flight Manual...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/07/2018</td>
<td>Active</td>
<td>10066169</td>
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<tr>
<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>REV. 1</td>
<td>EC036 T3 EC035 T1, EC035 T2+, EC035 T3, EC035 P3 EC035 T1, EC035 T2+, EC035 T3 EC135 T1, EC135 T2, EC135 T2+, EC135 P3, EC035 P2+, EC035 P3, EC035 P1, EC035 P2, EC035 P2+</td>
<td>Installation of Aerofrettcher MK IV Modular EMS System in accordance with the above listed document. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-27 amendment 2 forthe following paragraphs. 27.625/27.783(a)/27.807(a)(1)/27.1413(3) The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with FTI Engineering Network GmbH modification ref 18-1632 Master Data List F18- EO18-1632, Issue 01 dated 09.07.2018 and Installation Instructions ref F23-001- EO18-1632, Issue 2, dated. 06.07.2018. Operation in accordance with FTI Engineering Network Contact Aircraft Flight Manual...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/07/2018</td>
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<tr>
<td>SPECIALIST AVIATION SERVICESLIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>AW169</td>
<td>AW169 Paediatric Intensive Care Interior (SAS/MOD/935). This STC includes the installation of optional TCAA Litter Bridge (SAS/MOD/935).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Declaration of Compliance for SAS/MOD/935 Issue A, dated 11/07/2018 Rotorcraft Flight Manual Supplement FMS935-043 Issue A, Rotorcraft Flight Manual Supplement FMS935-043 Issue A Maintenance Program Supplement MP5/935-040 Issue A and later revisions of it.</td>
<td>NVIS operation requires separate design approval. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/07/2018</td>
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 insurers with this change
attributable to this certificate.
approval.

The Certification Basis (CB) for the original product remains applicable to this certificate.
approval.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:
Special Condition (s): CRI F-4312-001 Non-rechargeable lithium battery installations. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

Bell 505 VVIP Interiors P/N
6B5J0000X001001 in accordance with Design Data Set DDS-18-001 Rev. C.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

Bell 505 VVIP Interiors P/N
6B5J0000X001001 in accordance with Design Data Set DDS-18-001 Rev. C.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

Special Condition (s): CRI F-01, Security protection of aircraft systems/ networks. The following paragraph (s) at a later amendment: CS 25 Amendment 19C/CS25.603; CS25.788; T...

The Change is defined through AMAC Master Data List MDL-17099, Revision 1, dated December 19th, 2017, approved, and DECA Aviation Engineering Ltd. Instructions for Continued Airworthiness/CATI909, Revision 1, dated December 19th, 2017 (accepted) or later revisions of the above listed document(s) approved/accepted by the EASA.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

Special Condition (s): CRI F-01, Security protection of aircraft systems/ networks. The following paragraph (s) at a later amendment: CS 25 Amendment 19C/CS25.603; CS25.788; T...

The Change is defined through AMAC Master Data List MDL-17099, Revision 1, dated December 19th, 2017, approved, and DECA Aviation Engineering Ltd. Instructions for Continued Airworthiness/CATI909, Revision 1, dated December 19th, 2017, (accepted) or later revisions of the above listed document(s) approved/accepted by the EASA.

See those of original Transport Canada STC 17-121 Revision 1, dated December 19th, 2017, prior approved revisions. Maintenance is to be in accordance with DECA Aviation Engineering Ltd. Master Data List MDL-17099, Revision 1, dated December 19th, 2017, (accepted) or later revisions of the above listed document(s) approved/accepted by the EASA.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Limited to Helicopters having EASA STC 10017058 installed. (Starboard Meeker Utility Step with Fore C10 Mounting Arm for FLIR 380HD/3IPD payloads and All-A11 Mounting Arm for Trakka A800 payload). Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

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<th>EASA STC no.</th>
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<th>STC Ref.</th>
<th>Rev.</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10066238</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A320-232</td>
<td>REV. 1</td>
<td>LifePort, Inc. 20/07/2018</td>
<td>- Supplemental Type Design Document STC015-1745AM Rev0126: Aircraft Flight Manual AFM A-00-1854/ AFM Iss/Rev 0126 Instruction for Continued Airworthiness ICA A-00-18/54 ICA Iss/ Rev01450: Cabin Crew Data A-00-18/54/CD Iss 04/0r later revisions of the above listed document(s) approved/accepted</td>
<td>STC 10061321 for cabin completion needs to be installed before this design change. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/06/2019</td>
<td>Active</td>
<td>10066236</td>
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<td>10066248</td>
<td>GIVH AEROSPACE (IRELAND) LIMITED</td>
<td>A319-131, A320-232, A321-231</td>
<td>REV. 4</td>
<td>Installation of In-flight Canada USB, AES AC Power Systems and In-flight Canada Provisions Kit Rev. 1 extends the applicability by adding Airbus A319-131 A320-232 A321-231 (Rev. 2 extends the applicability by adding Airbus A320-232 A321-231) Rev. 3 updates the section of Description Design Change Rev. 4</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any change. In-flight Canada USB, AES AC Power Systems and In-flight Canada Provisions Kit Rev. 1 extends the applicability by adding Airbus A319-131 A320-232 A321-231 (Rev. 2 extends the applicability by adding Airbus A320-232 A321-231) Rev. 3 updates the section of Description Design Change Rev. 4. A320-232 A321-231</td>
<td>GIVH Aerospace Master Data List GIVH-8141-01-MDL-Q1 Issue 10(-) =&gt; GIVH Aircraft Flight Attendant Manual Supplement GIVH-8141-01-FAS-01 Issue 4(+) =&gt; GIVH A321-231 A320-232 A319-131 A320-232 A321-231</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/03/2021</td>
<td>Active</td>
<td>10066248</td>
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<td>10066260</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>FAA STC STO4219AT</td>
<td>A320-323/341/342/343</td>
<td>Version 7.1 Software upgrade of the Aviation Communication &amp; Surveillance Systems (ACSS) TCAS II Model TCAS 2000, P/N 751700-10200/Configuration 1 or Model TCAS, P/N 0000003-1414 (Configuration 2). This upgrade allows for the installation of Configuration 1 TCAS II. This installation also includes a</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List A330-15015-010, Revision O, dated 15 July 2015 older FAA approved revisions and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/07/2018</td>
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<td>10066260</td>
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<td>10066262</td>
<td>LIFEPORT, Inc.</td>
<td>FAA STC SA00153W</td>
<td>SEE FAA AML</td>
<td>Installation of Overhead Console Cabinets</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>LifePort, Inc., Master Drawing List B60-1132, Revision O, dated 25 June 2017, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/07/2018</td>
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<td>10066262</td>
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<td>10066265</td>
<td>AIRBUS S.A.S.</td>
<td>A321-231, A321-212, A321-211</td>
<td>MOD AXO10 ADDITIONAL WATER TANK INSTALLATION: Installation of an additional water tank (capacity 165l) in the lower part of cabin between F35.B and F36.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- MAS A00101 issue 01, dated 19/07/2015 =&gt; A32- A32-17132 issue 04 Certification Summary Report, dated 19/07/2018.</td>
<td>None</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the product. The STC is valid for aircraft types and models.</td>
<td>20/07/2018</td>
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<td>EASA STC no.</td>
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<td>Limitations/Conditions</td>
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<td>10066272</td>
<td>AVONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td></td>
<td>TB-M700 N, TB-M700 C1, TB-M700 C2</td>
<td>TB-M700 A, TB-M700 B</td>
<td>Installation of Comand CI 406-490.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: EASA CS 23, Amdt. 04, dated 15.07.2015: 23.230; 23.303; 23.305; 30.2307; 23.307(a); 23.337(a)(1); 23.365(a); 23.370(a)</td>
<td>Master Data List ASR-2017-006-MDL-01-00 Rev. 00 dated 13.07.2018; Airworthiness Limitation Section (§3) of ASR-2017-006-11-CA-00 Rev. 00 dated 19.03.2018 or later revisions of the above listed document(s) approved/accepted under the EASA System.</td>
<td>This STC covers the mechanical installation of the Comand CI 406-490 antenna only. Functional aspects must comply with the documentation of the connected devices.</td>
<td>23/07/2018</td>
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<td>10066273</td>
<td>AVCON INDUSTRIES, Inc.</td>
<td></td>
<td>FAA STC STO1811W</td>
<td>35, 35A, 36, 36A</td>
<td>Installation of the Garmin GTS 8000 TCAS II System and Dual GTX 3000 Transponders for ADS-B Out Compliance and Interfaced factory installed PC-530 Autopilot System to Single/Dual GTN 7XX/X7X Navgators.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Master Data List, Document No. 04701000, Rev. IR, dated 28 August 2019 or later FAA approved revision; and 2. Airplane Flight Manual Supplement, Document No. 04707100, Rev. 00 dated 5 July 2018, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/07/2018</td>
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<td>10066318</td>
<td>BIGORRE AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STO4265CH</td>
<td>MYSTERE FALCON 900 (MF900) FALCON 900EX (F900EX)</td>
<td>Installation of a Cabin LED Lighting System Rev. 1. Transfer from GDS AERO LLC. to BIGORRE AEROSPACE CORPORATION.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- GDS Master Data List (MDL); GDS-F900-002, Revision IR, dated 26 November 2016; Instructions for Continued Airworthiness, GDS- F900-030, Revision IR, dated 22 November 2016 for later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/02/2021</td>
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<td>10066327</td>
<td>PRATT &amp; WHITNEY ENGINE SERVICES, IN</td>
<td>REV. 2</td>
<td>FAA STC SA005580</td>
<td>PC-12/PC-12/45/ PC-12/47</td>
<td>Installation of the Pratt &lt;(&gt;&amp;&lt;) Whitney Engine Services FAST System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Pratt &amp; Whitney Engine Services Master Data List (MDL), Document number QM-T-100-B, Revision Q, dated 28 August 2019, or later FAA approved revision; Pratt &lt;(&gt;&amp;&lt;) Whitney Engine Services Aircraft Flight Manual (AFMS) Document number QM-T-200-B, Revision C, FAA approved on 27 September 2019&gt; or later FAA approved revision</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/05/2020</td>
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<td>10066333</td>
<td>CHIEFFEA AEROSPACE, Inc.</td>
<td></td>
<td>FAA STC STO411AT</td>
<td>550/ 560/ 560XL</td>
<td>Installation of Upgraded IC-615 with NZ-2000 FMS Software Version 6.1</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments or indicated on FAA STC. There are requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>1. Chieffea Aerospace Master Drawing List (MDL) No. 580-3461-01-001, Rev. IR, dated 28 December 2016, or later FAA approved revision; 2. Airplane Flight Manual Supplement (AFMS) to Model 550 (Brazi), 560 (Encore), 560XL (Excel), 560XL (XLS) for Upgrade for the IC-615 with NZ-2000 FMS Software Version 6.1.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/08/2018</td>
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<td>STA Code</td>
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<td>10066351</td>
<td>B/E AEROSPAC...</td>
<td>ST04427AT</td>
<td>Installation of Rockwell Collins Automatic Dependent Surveillance – Broadcast (ADS-B) Out System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Instructions for Continued Airworthiness (ICA) Document 2-191114-804, Aircraft Flight Manual Supplement (AFMS) Document 2-191114-805, or later revisions of the above listed document(s) approved/ accepted under the EASA in accordance with the Technical Implementation Procedures of E.U. USA Bilateral Agreement.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>21/12/2018</td>
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<td>10066352</td>
<td>ALOFT AEROARCHITECT S</td>
<td>ST03746NY</td>
<td>Vassali KA Cabin Distribution System Installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments as indicated on FAA STC. There requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. Master Data List (MDL) 1286000, revision 005, dated 23 May 2018, or later FAA approved revision for all above, activated, provisions and deactivated configurations. 2. Instructions for Continued Airworthiness (ICA) 1265316, revision 001 dated 06 January 2018, FAA accepted 24 May 2018 and later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>09/08/2018</td>
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<td>10066355</td>
<td>CARLISLE INTERCONNECT TECHNOLOGIES</td>
<td>ST04292CH</td>
<td>Installation of ARINC 791 Antenna Provisions (VT) for installation of ARINC 791 Antenna Provisions with the following configurations: 1. ARINC 791 SATCOM Antenna System Provision I.e. (Section 2.2), Installation of Emergency Locator Transmitter (ELT) Antenna Internal/Structural Support. 3. ARINC 791 SAT...</td>
<td>The Certification Basis for the original product as amended by additional or alternative airworthiness requirements of the Safety Standards.</td>
<td>Master Data List (MDL) 960-102885 Revision C dated 21 June 2018, or later FAA approved revision for all above, activated, provisions and deactivated configurations.</td>
<td>1) The equipment for which these provisions are intended has not been verified. Additional approval is required for the installation of this equipment and must be evaluated to assure satisfactory compliance with applicable airworthiness standards. 2) If the holder agrees to permit another person to...</td>
<td>06/08/2018</td>
<td>Active</td>
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<tr>
<td>10066362</td>
<td>TURK HAVA YOLLARI TEKNIK A.S (TURKISH TECHINC INC.)</td>
<td>A340-542</td>
<td>TC-CAN Meeting Room Modification</td>
<td>The Certification Basis for the original product as amended by additional or alternative airworthiness requirements of the Safety Standards.</td>
<td>Civil Aeronautical-CEP-04, Civil Aeronautical-CD01-01 and later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this design change/repair it must be determined that the interrelationship between this design change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>03/08/2018</td>
<td>Active</td>
<td>10066362</td>
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<td>10066363</td>
<td>TURK HAVA YOLLARI TEKNIK A.S (TURKISH TECHINC INC.)</td>
<td>A340-542</td>
<td>TC-CAN Monument Introduction</td>
<td>The Certification Basis for the original product as amended by additional or alternative airworthiness requirements of the Safety Standards.</td>
<td>Civil Aeronautical-CEP-04r0, or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this design change/repair it must be determined that the interrelationship between this design change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>03/08/2018</td>
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<td>10066364</td>
<td>AKV, Inc. (EASA.IM.R.S.00292)</td>
<td>AR14044LA</td>
<td>Installation of an AKV, Inc (formerly AKV Partners) Ns cycle counter/pn...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Drawing List No NCC-001, Rev. B, dated 10 January 2004, or later FAA approved revisions.</td>
<td>Prior to installation of this design change/repair it must be determined that the interrelationship between this design change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>13/08/2018</td>
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<td>365 AEROSPACE DDS LIMITED</td>
<td>365 AEROSPACE DDS LIMITED</td>
<td>10066398</td>
<td>REV 3</td>
<td></td>
<td>B300, B300C, B200, B200C, B200GT, B200CGT</td>
<td>Change to install a CVDR on Beechcraft Rev. 1: to include similar aircraft types Rev. 2: Transfer of STC from 365 Aerospace Ltd. to 365 Aerospace DDS Ltd</td>
<td>10066398</td>
<td>09/08/2018</td>
<td>03/07/2024</td>
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<td>MORSO PROJECTS LIMITED</td>
<td>MORSO PROJECTS LIMITED</td>
<td>10066395</td>
<td>REV 2</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>B200GT, B200CGT</td>
<td>Installation of two LITEF LCR-100N Attitude and Heading Reference Units (AHRS) on the B200GT/B200CGT equipped with Rockwell Collins Fusion (Change MP0004A).</td>
<td>10066395</td>
<td>09/08/2018</td>
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<td>54A SOLUTIONS FOR AVIATION, S.L</td>
<td>54A SOLUTIONS FOR AVIATION, S.L</td>
<td>10066419</td>
<td>REV 0</td>
<td></td>
<td>FALCON 10</td>
<td>INSTALLATION OF FLIGHT DATA RECORDER L3 SSFD FOR MODEL F1100</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>10066419</td>
<td>09/08/2018</td>
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The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

Condition: Documents issued with modification kits delivered before June 2019 remain valid and applicable unless expressly noted on later approval revisions.

Limitation: Modification section of AFM (Rev. 3) of this installation by third person is subject to written confirmation by the appr...
### List of EASA Supplemental Type Certificates
easa.europa.eu

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<th>Foreign/NAA STC ref.</th>
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<td>NORTWEST AEROESPACE TECHNOLOGIES INC.</td>
<td>10066428</td>
<td>REV. 2</td>
<td>777-300</td>
<td>Interior Reconfiguration of Cathay 777-300 Aircraft with Optional Installation of Bucher Stretcher.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition (s): CRI F-01 (Non-rechargeable Lithium Battery Installation(s)), CRI-D-01 (Inflatable Restraint (Systems)): The following paragraph(s) at a later amendment...</td>
<td>- Certification Basis Summary: Northwest Aerospace Technologies, Inc. document NO593-D00, Revision G, dated 17 May 2019: - Instructions for Continued Airworthiness, Northwest Aerospace Technologies, Inc. document NO593-D70, Revision C, dated 8 August 2018 for the FAA STC and prior to installation of any optional items.</td>
<td>22/05/2019</td>
<td>Active</td>
<td>10066429</td>
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<td>WEST STAR AVIATION, LLC/ bia WEST STAR AVIATION</td>
<td>10066443</td>
<td>REV. 1</td>
<td>FAA STC ST00818DE</td>
<td>Installation of Honeywell Control Display Unit (CDU) model CD-830 to replace the following CDU's: CD-810, CD-815 and CD-820. Revision 1: Addition of Textron Cessna 560 XL.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25.25, 25.77(1)(a)(i), 25.77(1)(a)(ii), 25.777(a)(b)(c), 25.1301 (a)(1)(x)(2) (j), 25.1309 (a)(3)(c)(ii), 25.1330, (...</td>
<td>- EASA Approval of FAA-PMA (Part Manufacturing Approval) approved.</td>
<td>22/02/2023</td>
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<td>10066443</td>
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<td>AMETEK AMERON, LLC/ B&amp;W MASS SYST</td>
<td>10066444</td>
<td>REV. 1</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>EASA Approval of FAA-PMA (Part Manufacturing Approval) approved. Reassembly – Door Emergency Power Plant Nº 40452.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- FAA PMA No. PQ1432NM Supplement No. 1, dated 27 July 2013: and amendment 26 December 2012, - Technical documentation supersedes the Approved Model List (AML) for the EASA STC 10066444 or revisions of the above listed document(s) approved on behalf of EASA in accordance with the Technical Implementation</td>
<td>11/06/2018</td>
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<td>STG AEROSPACE GERMANY GmbH</td>
<td>10066445</td>
<td>REV. 1</td>
<td>A321-231, A321-232, A321-233, A321-234</td>
<td>AIRBUS LITEMOOD® CABIN LIGHTING UPGRADE:Replacement of the Airbus Single Axle fluorescent ceiling; window, entry and lavatory lighting with new STG Aerospace Airline LithiMod® LED Lighting. Revision 1: Transfer of STC holder from STG Aerospace Limited to STG Aerospace GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List SG-1024MDL-01 Rev. IR, dated 24 Jul 2018: - Additional Limitations and Conditions may be listed on AML for the EASA STC 10066444,</td>
<td>12/02/2021</td>
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List of EASA Supplemental Type Certificates

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<td>10068465</td>
<td>SWIS CERTIFICATION SERVICES LTD.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>SkyCouch Additional Occupancy</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 6. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
<td>Master Configuration Scheme, Ref. 3166- MCS-01 Issue 2 or later revisions of the above listed documents approved/ accepted under the EASA system.</td>
<td>- SkyCouch can only be used as a sleeping or an accommodation area in certain flight conditions, as detailed in SWIS Certification Services Ltd Design Report 3166-DAR-01.</td>
<td>20/02/2018</td>
<td>Active</td>
<td>10068465</td>
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<td>10068467</td>
<td>LEE AEROSPACE, INC.</td>
<td>FAA STC ST0162W</td>
<td>DHC-8-301/311-315-401</td>
<td>DHC-8-102/103-106-201</td>
<td>DHC-8 Window Assembly Side Canopy (Rt) Replacement Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/revised. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List Doc. No. LA/ADS-152, Rev C, dated 19 January 2018 or later revisions of the above listed documents/approved/ accepted under the EASA System.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse affect upon the airworthiness of the product.</td>
<td>15/11/2019</td>
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<td>10068469</td>
<td>BHE &amp; ASSOCIATES LTD.</td>
<td>FAA STC ST121/00S</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of automatic dependent surveillance-broadcast (ADS-B) outon Part 25 aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. There are no requirements for environmental protection and the associated noise and emissions levels of the original product.</td>
<td>Master Data List 584-03-0001, Revision IR, dated 29 June 2018 or later revisions of the above listed documents approved/ accepted under the EASA System.</td>
<td>1. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse affect upon the airworthiness of the product.</td>
<td>03/06/2019</td>
<td>Active</td>
<td>10068469</td>
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<tr>
<td>10068473</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>737-400</td>
<td>Palletized Aerial Dispensing of Liquids</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 14 except for Electrical Wiring Interconnection System complies with ...</td>
<td>EASA Approved WPA Aircraft Flight Manual Supplement ref. WPA-5300-AMBS-14, Revision C, dated 15 August 2018 WPA Instructions for Continued Airworthiness ref. WPA-5300-IA-14, Revision 0, dated 12 May 2018 EASA Accepted WPA Master Data List WPA-5300-MDL-14, Revision A, dated 12 May 2018 or later revisions of ...</td>
<td>This STC can only be installed on a Boeing 737-400 series airplane modified in accordance with Aeronautical Engineers, Inc. FAA STC ST16275A (valid for EASA under STC ref. 10015732 Rev 4 or later revisions), and Ventura Aerospace Inc. cargo barrier FAA STC ST1806 (validated by EASA under STC ref. 10068473).</td>
<td>20/12/2019</td>
<td>Active</td>
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<td>10068482</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A319-131, A319-132, A319-133, A319-111, A319-112, A319-113</td>
<td>A319 Cargo Access Hatch Installation of new maintenance access hatch at C61-C62 from cabin toafh bulk cargo compartment.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: EC 2324/08 Enhanced Airworthiness Programme for Airc...</td>
<td>- Master Document List P5- AA-E01-MDL-0001 Issue A, dated 16/08/2018, Instruction for Continued Airworthiness (ICA) ALS P5-44401-ICA-000 Issue A, dated 16/08/2018 or later revisions of the above listed documents/approved/ accepted under the EASA System.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse affect upon the airworthiness of the product.</td>
<td>17/06/2018</td>
<td>Active</td>
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<td>10068512</td>
<td>JET AVIATION AG</td>
<td>REV. 1</td>
<td>767-300</td>
<td>KA Band Antenna Installation.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 6. The requirements for environmental protection and the associated noise and emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>For configuration 1: JBSL- M-00776AFM-01 Issue 1 or for configuration 2: JBSL- M-00776AFM-01 Issue 1 For configuration 3: documents from configuration 1 and 2 or later revisions of the above listed documents approved by EASA.</td>
<td>Configuration 2 may be installed with the structural provisions defined in Configuration 1 and any other EASA approved structural provision (see JBSL-M-00776ANA-01 Issue 1). Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other...</td>
<td>28/06/2018</td>
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<td>EASA STC no.</td>
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<td>Foreign/NAA STC ref</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10068515</td>
<td>JORMAC AEROSPACE, INC.</td>
<td>FAA STD ST0244SAT</td>
<td>777-200, 737-700/800-900/900ER</td>
<td>Installation of External Door, Access Panel and Window Security Locks</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Jormac Aerospace Master Drawing List JNER-843, Revision AC, dated 16 July 2015 or later FAA approved revision 2. Airplane Flight Manual Supplement, JNER-847, Revision L, dated 19 February 2016, or later FAA approved revision 3. Instructions for Continued Airworthiness, JNER-865, Revision J, dated 19 February 2016.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/06/2018</td>
<td>Active</td>
<td>10068515</td>
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<tr>
<td>10068566</td>
<td>P3 ENGINEERING GMBH</td>
<td>A330-341, A330-342, A330-343, A330-321, A330-322, A330-333, A330-301, A330-302, A330-303, A330-223, A330-243, A330-202, A330-203</td>
<td>P3 Engineering GmbH EON 150050, Rockwell Collins Tailwind TV9580, TVSystem Removal, This modification de-installs the system components and the fuselage-mounted radome and antenna located in between frames C37 and C37.4.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>P3 Engineering GmbH Instructions for Continuing Airworthiness ref. ICAP-AAA01-ICA-0001 (Issue A)</td>
<td>The requirements for the above listed document(s) approved/ accepted/derivative of the EASA system.</td>
<td>Aeroplanes modified by this STC must be maintained in accordance with P3Engineering GmbH Instructions for Continuing Airworthiness ref. ICAP-AAA01-ICA-0001 document.</td>
<td>24/08/2018</td>
<td>Active</td>
<td>10068566</td>
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<tr>
<td>10068571</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A321-231</td>
<td>VIP Cabin Completion</td>
<td>The Certification Basis for the original product as amended the following additional or alternative environmental requirements the following paragraph(s) at a later amendment: CS 25.663, 25.788 (b), 25.10 (a), 525.50 (a), 525.50 (a), 525.50 (a) at Amendment 10.</td>
<td>Supplemental Type Definition Document (STDD) Ref. STDD00-1701M Issue 01 Revision 00 Compliance Checklist (CLC) Ref. CRL00-1701M Issue 01 Revision 00 and CRL00-1801M Issue 01 Revision 00 Airplane Flight Manual Supplement (AFM-S) Ref. A-400-17122-01AFM Issue 00-00Instructions for Continued Airworthiness.</td>
<td>The aircraft is certified for &quot;private operation&quot; only. In the cabin configurations &quot;empty space for one or two patient transport units&quot; the opposite seats (left side: left side of cabin) of the empty space must not be occupied. As defined in Section 4 (ALS) of the Instructions for Continued Airworthiness.</td>
<td>24/08/2018</td>
<td>Active</td>
<td>10068571</td>
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<tr>
<td>10088361</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>767-300</td>
<td>INTRODUCTION OF Iridium SATCOM</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>MASTER DOCUMENT LIST: S21 MEL-1397 ISSUE 2 AIRCRAFT FLIGHT MANUAL SUPPLEMENT S21 FMS-0144 ISSUE 1, later revisions of the above listed document(s) approved/accepted/derivative of the EASA system.</td>
<td>The SATCOM system must not be used during take-off/landing or approach. The SATCOM system is approved for non-essential voice communications only. It is not approved for Future Air Navigation System (FANS) or FidUrl Range Voice Communication with ATC. Prior to installation of this change/repair it must...</td>
<td>28/08/2018</td>
<td>Active</td>
<td>10088361</td>
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<tr>
<td>10088846</td>
<td>AVIANOR INC</td>
<td>A330-343</td>
<td>Cabin reconfiguration, seat installation and layout of passenger/accommodation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Avianor Master Document list MDL16162002 Rev. B dated 1 March 2018; - Avianor Master Document list MDL18182702 Rev. NC dated 18 February 2018; - Avianor Supplemental Instructions for Continued Airworthiness/CA16162002 Rev. A dated 1 March 2018 or later revisions of the above listed documents approved/accepted/derivative of the EASA system.</td>
<td>Limited to s/n 1098Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/06/2018</td>
<td>Active</td>
<td>10088846</td>
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</table>
For Configuration 1: The change is defined with MasterData List MDL B7472-1-5300 Revision D.

- The aircraft shall be operated in accordance with Airplane FlightManual Supplement APAMS B747-3-1-5300 Revision A.
- The aircraft shall be maintained in accordance with Instruction for Continued Airworthiness.

The EASA Approval is only applicable Airplanes FALCON 900B MBB 38, 73, 74, 77 and 90. Before accomplishment of this modification Transponder MESTERB must be installed with ADS-B-NDA capabilities as per modification SIRUMAERO TECH SAT17-61.

- EASA Major Change 10065263 Rev. 1 embodiment is a pre-condition - Limited to helicopters with Helionix cockpit version.

Prior to accomplishment of this change/repair it must be determined that the associated technical interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Prior to installation of this change/repair it must be determined that the associated technical interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.
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<tr>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>-</td>
<td>DORNIER 228-122</td>
<td>Approval of Single / Dual Honeywell MTE 70B Transponder system, EHS &amp; ADS-B upgrade and approval of Single / Dual Rockwell Collins TDR-940/Transponder system, EHS &amp; ADS-B upgrade, in accordance with Scandinavian Avionics modification 2111499-EO. Issue 2</td>
<td>The Certification Basis for the original product as amended by adding two or more additional or alternative airworthiness requirements. Deviation(s): C5, C5-AC56, A23, A23-AC56. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>- Modification is in accordance with Scandinavian Avionics DesignEngineering Order 2111499-EO issue 2, dated 24.06.2019 - Operate in accordance with Scandinavian Avionics Design Flight Manual Supplement 2111499-AM A Issue 2- (&gt;), (&lt;) dated 08.05.2019 and 2111499-AM Clause 4, dated 17.06.2019-</td>
<td>21/06/2019</td>
<td>Active</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>FAA STC ST06871AC</td>
<td>737-800</td>
<td>Installation of headliner hanger and panel system.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments as indicated on FAA STC ST06871AC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>1. Bullet list 144600001-001 revision AR dated 1 June 2017 or later FAA approved revision. 2. Bullet list for FA continued Airworthiness Instruction(s) R115250001 Rev R dated 19 September 2016 or later FAA approved revision.</td>
<td>Prior to installation of this design change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will not introduce adverse effect upon the airworthiness of the product.</td>
<td>04/04/2019</td>
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<td>10066729</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SA09744DS</td>
<td>N25/ P35/ S35H35/ J35S/ K35/ M35/ V35/ V35A/ V35B/ 36/ 361/ 361/ 361/ 361/ 361</td>
<td>Installation of an S-TEC 01326 Digital Flight Control System (DFCS).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments as indicated on FAA STC SA09744DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>1. Master Data List No. ST-959-ML-0001, Revision A, dated 06 April 2018 or later FAA approved revision 2. Airplane Flight Manual Supplement, ST-959-AMS-0001, Revision A dated 06 April 2018 or later FAA approved revision 3. Instructions for Continued Airworthiness, ST-959-ICA-0001, Revision A dated 30 October 2017.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and other design change/repair it must be determined that the interrelationship between these design change/repair introduces no adverse effect upon the airworthiness of the product.</td>
<td>06/09/2018</td>
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<td>10066732</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 1</td>
<td>FAA STC SA09681J</td>
<td>Boeing 737-800 T-PED (Transmitting Personal Electronic Device) tolerance demonstration.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Social Air France Airplane Flight Manual Supplement/AMS- B737-18-001-001 original issue/ Social Air France Instruction(s) for Continued Airworthiness of Aircraft T-PED Tolerance (ICA)- B737-18-001-001 original issue or later revisions of the above listed document(s) approved/accepted under the EASA system...</td>
<td>The modification consists in demonstrating that B737-800 aircraft T-PED tolerant to mobile phone and Wi-Fi frequencies. Transmitting PEDs can be used in passenger cabin in transmitting mode during all phases of flight but restricted to connections as per cellular and/or wireless technologies. The use...</td>
<td>05/02/2018</td>
<td>Active</td>
<td>10066734</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10066758</td>
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<td>MD-11F</td>
<td>Installation of PPG Aerospace main cockpit windshields P/N 196700-1Stand 196700-02</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unmodified and remain applicable to this certificate/approval.</td>
<td>- Supplemental Type Design Document STC/DOA-1803M, Issue/Rev 1/0; - Compliance Check List CC/34-1803M Issue/Rev 2/0, or later revisions of the above listed document(s) approved.</td>
<td>Prior to installation of this design change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2018</td>
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<td>10066792</td>
<td>SIDDHIS AERONAUTICAL DESIGNSCORPORATION</td>
<td>REV. 2</td>
<td>TCCA SA/89-92 ISSUE 7</td>
<td>CL-600-2B19 (RJ SERIES 440) CL-600-2B19 (RJ SERIES 100) CL-600-2B16 (VARIANT) CL-600-2B16 (601-3R VARIANT) CL-600-2B16 (601-3A VARIANT) CL-600-2A12 (601 VARIANT) CL-600-2A11 (600)</td>
<td>Installation of fin-mounted radome and SatCom / DBS antenna (Structural approvals only).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unmodified and remain applicable to this certificate/approval.</td>
<td>The change installation must be performed in accordance with DECA Aircrane Engineering Limited modification summary document number MS15040, Revision 7, dated 15 January 2019. Operation must be performed in accordance with DECA Aircrane Engineering limited flight manual supplement number FM610046, Revision...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/10/2019</td>
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<td>10068800</td>
<td>JETTECH, LLC.</td>
<td>FAA STC SD/432LA</td>
<td>501/551</td>
<td>New Garmin GTN 6XX/7XX Series Global Navigation Systems with WAAS capability.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unmodified and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unmodified and remain applicable to this certificate/approval.</td>
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<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/09/2018</td>
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<td>10068802</td>
<td>JETTECH, LLC.</td>
<td>FAA STC SD/1755W</td>
<td>525, 525A, 525B</td>
<td>Installation of GPS LFRIGHT Deviation Converter with a single or dual GTN 6XX/7XX GPS WAAS receiver to allow the installed AFCS to be WAAS LPR capable.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unmodified and remain applicable to this certificate/approval.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unmodified and remain applicable to this certificate/approval.</td>
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<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/09/2018</td>
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<td>10068827</td>
<td>PPG AEROSPACE</td>
<td>FAA STC ST/059HLA</td>
<td>400 400A</td>
<td>Installation of PPG Aerospace main cockpit windshields P/N 196700-1Stand 196700-02</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unmodified and remain applicable to this certificate/approval.</td>
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<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/09/2018</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>FUNT AERO, INC.</td>
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<td>FAA STC</td>
<td>A185E/ A185F/ 185/ 185A/ 185B/ A185/ 185D</td>
<td>Installation of wing tip fuel tank 28/30 U.S. gallon; increase wing span approximately four foot and area to 185 square foot and increase gross weight to 3,800 lbs. for laminates.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>First Master DRAWING LIST (MDL) FA 185. Revision L, dated 29September 1995, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2018</td>
<td>Active</td>
<td>10686823</td>
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<tr>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td></td>
<td>EASA STC</td>
<td>S4A008NM</td>
<td>B777 PM-CPDLC ACARS CMU II UPGRADE Installation of Honeywell/ACARS Mark II CMU and associated wirings and changes. The target aircraft for this STC are currently provisioned for ACARS CMU’s and will be upgraded to modify the wiring configuration to facilitate installation of the Mark II Honeywell 955-075B-006 AC.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (v) OR F-04 Flight Recorders and Data Link Recording, following paragraph(s) at a later amendment: CS-ACNS Initial Issue. The requirements remain unchanged and remain applicable to this certificate/approval.</td>
<td>MASTER DOCUMENT LIST ISSUE 120 later revisions of the above listed document(s) approved/accepended under the EASA system.</td>
<td>Prior to installation of this STC, the authorizing design authority shall determine the interrelationship between this change and any other previously installed design change and repair.</td>
<td>11/09/2018</td>
<td>Active</td>
<td>10686845</td>
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<td>CAA UK RESPONSIBILITIES</td>
<td></td>
<td>EASA STC</td>
<td>A321-213/221-232</td>
<td>Introduction of ARINC 791 SATCOM System. Introduction of the SATCOM components of an ARINC 791 satellite communications system in an existing external adapter plate and radome.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master document list: S21 MDL-14011+/−. Prior to installation of the above listed document(s) approved/accepended under the EASA system.</td>
<td>A radar and VSAT adapter provisions must be installed prior to occurrence with this design change. The provisions must be compatible and have been previously substantiated to embody a SATCOM Antenna with the following characteristics: Weight: 60Kg; Swept Diameter: 980mm; Height: 280mm. Prior to installation of this STC, the authorizing design authority shall determine the interrelationship between this change and any other previously installed design change and repair.</td>
<td>12/09/2018</td>
<td>Active</td>
<td>10686846</td>
</tr>
<tr>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td></td>
<td>EASA STC</td>
<td>A320-211/220-214/215-216</td>
<td>B777 Removal of Economy Class Seats: This Change removes all existing Economy Class passenger seats in the main cabin of the aircraft without replacement to achieve a First Class business class only configuration for limited ferry flights. Another element in the passenger cabin will remain as is...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-25 Amendment 21 relate Certification Basis for this project. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List: MDL/1214-001 Issue 3. Flight Manual Supplement FMS/1214-001 Issue 2. Prior to installation of the above listed document(s) approved/accepended under the EASA system.</td>
<td>Operation of the aircraft is restricted to non-revenue ferry flight with crew and supervisory crew only. The carriage of passengers is not permitted. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/09/2018</td>
<td>Active</td>
<td>10686864</td>
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<tr>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td></td>
<td>EASA STC</td>
<td>PC-12/45</td>
<td>HF radio installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Operation in accordance with document: No. PC-1201-HDK01, Supplement/Manual Vuelo, Ed. 2, dated 04/08/2015. Installed and maintenance in accordance with document: No. PC-1201-HDK-02 Supplements Manual Maintenance, Ed. 6, dated 11/06/2016. Installation of the above listed documents approved by EASA.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this design change and any other previously installed design change and repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/09/2018</td>
<td>Active</td>
<td>10686867</td>
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<tr>
<td>ADVANCED AEROSPACE DEVELOPMENTS GMB</td>
<td></td>
<td>EASA STC</td>
<td>AS 350 D AS 350 B1, AS 350 B1, AS 350 B2, AS 350 B3</td>
<td>External Mirror System (EMS) consisting of two mirrors with electromechanical control and de-fog.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-27 Am14 4 for the affected parts. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List: AAD-MDL-016MS-001 Rev 04, dated 10 September 2018. Aircraft Flight Manual Supplement AAD- FMS-016MS-001 Rev 01, dated 19 June 2018. Installation Manual and Instruction for Continued Airworthiness AAD-ICA-016MS Rev 01, dated 27 June 2018. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/09/2018</td>
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<td>10686894</td>
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<td>STC Holder</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>A319-115(L)</td>
<td></td>
<td></td>
<td></td>
<td>CPDLC (Controller Pilot Data Link Communications) System Activation.</td>
<td>Société Air France S.A. Airplane Flight Manual Supplement/AMEC-080-000-001; Société Air France S.A. Flight Crew Supplement/COMS-080-000-001</td>
<td>- This STC does not, by itself, constitute an operational approval whenever an approval is required. The following functions are not certified: - Pushback clearance request&lt;=&gt; - Oceanic clearance request&lt;=&gt; - Taxi clearance request&lt;=&gt; - TWIP (Terminal Weather Information for Pilots) cl...</td>
<td>18/02/2018</td>
<td>Active</td>
<td>10066912</td>
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<tr>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A319-133</td>
<td></td>
<td></td>
<td></td>
<td>Certification for Commercial Operation / This change includes instructions to alter the aircraft to allow operation in Commercial Air Transport. If desired, this change also includes instructions to alter the aircraft back to its initial configuration for sole private use.</td>
<td>FOCAC approved STC Z25-20-72 and STC 10039943 has to be installed prior or in conjunction with the modifications of this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair is...</td>
<td>19/06/2018</td>
<td>Active</td>
<td>10066922</td>
<td></td>
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<tr>
<td>O.C.M. DESIGN GmbH</td>
<td>560</td>
<td></td>
<td></td>
<td></td>
<td>Installation of ADS-B Out and FMS Upgrade with LPV Approach Capability.</td>
<td>Installation in accordance with OCM Design Organisation modification refC60X-QCM-001 and Engineering Bulletin EB-C60X-QCM-001, Rev. 0 dated 15/07/2018, Operation in accordance with OCM Design Organisaton Aircraft Flight Manual Supplements-C60X-QCM-1001 Aviation, FAA approved IR dated 23-Jul-2018;...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effects upon the airworthiness of the product.</td>
<td>21/09/2018</td>
<td>Active</td>
<td>10066950</td>
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<tr>
<td>SABCOCK MISSION CRITICAL SERVICESDESIGN N AND COMPLETIONS LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td></td>
<td>MB-BK117 C-2, MBB-BK117 D-2</td>
<td>Investigation to demonstrate that MBB-BK117 D-2 and MBB BK117 C-2 are PED's and T PED's tolerant.</td>
<td>Babcock FMS 110-003 iss 2 for MBK-BK117 C-2 and MBK-BK117 D Babcock Design Report 101-002 iss. 3Babcock Compliance Checklist 101-004 iss 2 and later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>None</td>
<td>05/10/2018</td>
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<td>10066960</td>
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<td>Limitations/Conditions</td>
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<tr>
<td>10066962</td>
<td>PHOENIX AEROSPACE LIMITED</td>
<td>CAA UK RESPONSIBIL</td>
<td>212</td>
<td>Installation of a Garmin GTN650H.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS-ACNS Initial Issuance apply to the affected requirements. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certification.</td>
<td>Doc. No PA-4760-212-139 FLIGHT MANUAL SUPPLEMENT Revision 4(&gt;)</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>24/02/2018</td>
<td>Active</td>
<td>10066962</td>
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<tr>
<td>10066978</td>
<td>BOEING IRELAND LIMITED</td>
<td>REV. 2</td>
<td>777-300ER</td>
<td>Installation of a Garmin GTN650H.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS-ACNS Initial Issuance apply to the affected requirements. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certification.</td>
<td>Continental DataGraphics Ltd Aircraft Flight Manual Supplement No B1053-004A Revision 1(&gt;)</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>16/11/2020</td>
<td>Active</td>
<td>10066978</td>
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<tr>
<td>10066987</td>
<td>VT DRB AVIATION CONSULTANTS</td>
<td>REV. 1</td>
<td>FAA ST11214DS</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of a Garmin GTN650H.</td>
<td>FAA Approved Model List for Installation of a Garmin GTN650H.</td>
<td>FAA Approved Model List for Installation of a Garmin GTN650H.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>23/04/2019</td>
<td>Active</td>
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<td>10067004</td>
<td>JANCO AMERICA, INC.</td>
<td>FAA STC ST01805E</td>
<td>777-300</td>
<td>Installation of an interior reconfigurati</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS-ACNS Initial Issuance apply to the affected requirements. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certification.</td>
<td>Jamco America, Inc. Master Drawing List (MDL) 1343-1 Revision J dated 25 October 2013, or later Federal Aviation Administration (FAA) Supplemental Instructions for Continued Airworthiness (ICA) Document Number ICA1343-1, Revision K, dated 10 February 2012, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>26/09/2018</td>
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<tr>
<td>10067014</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA ST01177N</td>
<td>EMB-145MP, EMB-150ER EMB-145, EMB-145ER EMB-150SL, EMB-150BJ</td>
<td>Installation of ADS-B with Freewill GPS.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS-ACNS Initial Issuance apply to the affected requirements. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certification.</td>
<td>FAA Approved Model List for Installation of ADS-B with Freewill GPS.</td>
<td>FAA Approved Model List for Installation of ADS-B with Freewill GPS.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>27/08/2018</td>
<td>Active</td>
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</table>

**Certification Basis:** The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements.

**Limitations/Conditions:** Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.
<table>
<thead>
<tr>
<th>STC Holder</th>
<th>FAA STC No</th>
<th>Description</th>
<th>Certification basis</th>
<th>Foreign/NAA STC ref</th>
<th>Certification basis</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA ST0522SE</td>
<td>HGS 6000 on Boeing 737-NG</td>
<td>Installation of HGS Model 6000 on Boeing 737-NG aircraft. HGS6000 is a re-host of the HGS Model 4000 approved for installation on the Boeing 737-NG aircraft.</td>
<td>737-400ER 737-900 737-800 737-700</td>
<td>1. Certification Basis for the original product as amendability of the following additional or alternative environmental noise and/or emissions levels of the original product remains applicable to this certificate/approval. Installation of Cable Cutters.</td>
<td>Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/10/2018</td>
<td>Active</td>
<td>10067030</td>
</tr>
<tr>
<td>CHIPPENHA AEROSPACE, Inc.</td>
<td>FAA STC ST04190AT</td>
<td>Installation of ADS-B Out</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EMB-145SR, EMB-145M</td>
<td>1. Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/10/2018</td>
<td>Active</td>
<td>10067045</td>
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<tr>
<td>DART AEROSPAC USA, Inc.</td>
<td>FAA STC SR02605SE</td>
<td>Installation of Cable Cutters</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EC 130 T2 EC 130 B4</td>
<td>1. Installation Instructions (IN)-D130-1046, Revision B, dated 26 June 2018, or later FAA approved revision. Aircraft Equipped with Global Vision Flight Deck (GVFD) are excluded from this STC. Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/10/2018</td>
<td>Active</td>
<td>10067052</td>
<td></td>
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<tr>
<td>Q.C.M. DESIGN GmbH</td>
<td>BD-700-1A10</td>
<td>Honeywell CNX-900 WiFi Installation</td>
<td>Certification Basis for the original product as amendability of the following additional or alternative environmental noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Change Definition: Chapter 3.4 of Certification Program, Doc. No. CP-1.</td>
<td></td>
<td></td>
<td></td>
<td>02/10/2018</td>
<td>Active</td>
<td>10067056</td>
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<tr>
<td>PMV ENGINEERING</td>
<td>A321-273NX A321-253NX-271N-272N-251N</td>
<td>CVR/DFDR ULB 90 days</td>
<td>The aim of this modification is to replace the existing 30 days underwater locator beacon (ULB) installed on Cockpit Voice Recorder (CVR) and Flight Data Recorder (FDR) by new 90 days ULB.</td>
<td>A321-213/211-241/242-231N</td>
<td>CVR/DFDR ULB 90 days</td>
<td>Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/10/2018</td>
<td>Active</td>
<td>10067059</td>
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### List of EASA Supplemental Type Certificates

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<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA ref.</th>
<th>Model</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>10067069</td>
<td>GAMA AVIATION (ENGINEERING) LIMITED</td>
<td>CAA UK</td>
<td>RESPONSIBILITY</td>
<td>C90A, C90GT, C90GTI</td>
<td>Antenna Provision for Microwave D/L System.</td>
<td>The Certification Basis (CB) for the original product remains applicable: this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>GAEL-1635-QD-D1, issue 1, Dated 20th September 2019(&lt;{}&gt;) or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/10/2018</td>
<td>Active</td>
<td>10067069</td>
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<tr>
<td>10067075</td>
<td>AVIONTEC AG</td>
<td>PA-46-350P</td>
<td>Installation GPS WAAS Antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable: this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Installed and Maintained in accordance with the documents listed in Master Document List: Master Document List 8934-017, Revision 0, Dated 13th September 2018.</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/10/2018</td>
<td>Active</td>
<td>10067075</td>
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<tr>
<td>10067072</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC</td>
<td>50913XRC-D</td>
<td>AS 350 B3 AS 350 B2</td>
<td>Installation of a Lead Acid Battery RS350 at the Aircraft StandardBattery Location</td>
<td>The Certification Basis (CB) for the original product remains applicable: this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Drawing List 350A82-1753, initial release, dated 19 October 2000 or later FAA approved revision(s).</td>
<td>The modification ref. AMS 07-2709 is a pre-requisite for the embodiment of this Supplemental Type Certificate (STC) on AS 350 B2. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/01/2019</td>
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<td>10067076</td>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FAA STC</td>
<td>ST0392NY</td>
<td>FALCON 900EX (F900EX)</td>
<td>Installation of a Meggitt ISFD Functional Upgrade (Displays,Ls and Magnetic Heading) In Dassault Aviation 900EX Aircraft with EASyAvionics</td>
<td>The Certification Basis for the original product remains applicable: this certificate approval, except where amended by additional or later amendments if indicated on FAA STC ST0392NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>1. Dassault Aircraft Services Master Data List FWXX-DD033-006 Rev IR, dated 31 May 2018, or later FAA approved revision 2. Instructions for Continued Airworthiness (ICA) FWXX-DD083-155, Rev R, dated 16 May 2018(&lt;{}&gt;) or later FAA approved revision 3. Airplane Flight Manual Supplement (AFMS) FWXX D00.</td>
<td>This alteration requires the following prerequisites: TII-F900EX.XM5716-M0920 and equivalent mod. MBK Part Number 7028560-921 and MRC Part Number 7517959-904 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/10/2018</td>
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<td>10067083</td>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FAA STC</td>
<td>ST0359NY</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of Honeywell Laser/ IV Upgrade</td>
<td>The Certification Basis for the original product remains applicable: this certificate approval, except where amended by additional or later amendments if indicated on FAA STC ST0359NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>1. Dassault Aircraft Services Master Data List AML25-D0074-005, Revision A, dated 1 September 2017 or later FAA approved revision 2. Dassault Aircraft Services Instructions for Continued Airworthiness AML25-D0074-155, Revision R&lt;{}&gt;, dated 28 July 2017 or later FAA approved revision(s).</td>
<td>Prior EASA Approved installation of Honeywell Laser II, III and/or IVRS as specified by AML No. AML25-D0074-002 General Note 3 is a prerequisite to this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/10/2018</td>
<td>Active</td>
<td>10067083</td>
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<tr>
<td>10067084</td>
<td>AIRLIFT A/S</td>
<td>FAA STC</td>
<td>155515NY</td>
<td>AS 355 F2, AS 355 N, AS 355 F, AS 355 F1</td>
<td>The mod DO-355-25-101 introduces the &quot;Removable External Mount&quot; on AS355 series helicopters.</td>
<td>The Certification Basis (CB) for the original product remains applicable: this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>CCPL-355-25-101 Configuration Control Part List, rev 0, dated 01 June 2018/ FMS-355-25-101-1 Flight Manual Supplement, rev 04, dated October 2018/ICA-355-25-101-01 Instructions for Continued Airworthiness, rev 0, dated 01 November 2016 or later revisions of the above listed document(s) approved/accepted.</td>
<td>The external load installation and the task specialist restraint systems are not part of this STC and must be subject of separate approval. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/10/2018</td>
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<td>L2 CONSULTING SERVICES Inc.</td>
<td>777-200/777/300/777-300 ER</td>
<td>Installation of an Indium Communication System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. There are requirements for environmental protection and the associated certified noise level.</td>
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<tr>
<td>AVIONIK STRAUSSING ENTWICKLINGS GmbH</td>
<td>525, 525A, 526B</td>
<td>Installation of dual Garmin G*GPS/</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>BELL HELICOPTER TEXTRON Inc.</td>
<td>505</td>
<td>Installation of Headliner.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. There are requirements for environmental protection and the associated certified noise level.</td>
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<tr>
<td>TEXTRON, Inc.</td>
<td>M137, M332, M337 Engines Overhaul</td>
<td>The change AHUK/350/3143 introduces an Air Data Boom to be used independently of air data for special training need.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>BELL HELICOPTER TEXTRON Inc.</td>
<td>505</td>
<td>Installation of Tinted Windows.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. There are requirements for environmental protection and the associated certified noise level.</td>
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<tr>
<td>BELL HELICOPTER TEXTRON Inc.</td>
<td>505</td>
<td>Installation of Headliner.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. There are requirements for environmental protection and the associated certified noise level.</td>
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# List of EASA Supplemental Type Certificates

<table>
<thead>
<tr>
<th>STC Holder</th>
<th>STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>LUTTHANSU TECHNIK AG</td>
<td>REV. 1</td>
<td>MD-11F</td>
<td>Activation of ADS-B and ELT/EHS upgrade. The change consists in activating Automatic Dependent Surveillance – Broadcast Out (ADS-B Out) and to upgrade the Mode S Elementary Surveillance (ELS) and Enhanced Surveillance (EHS) System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unannounced and remain applicable to this certificate/ approval.</td>
<td>Supplemental Type Design Document: Doc. no. ST003A-1505M Issue/Rev 1/2, Manuals - Instruction for Continued Airworthiness: Doc. no. A-00-18/1-1/CASA/Rev 1/0, - Flight Crew Information: Doc. no. A-00-18/1-F02 Issue/Rev 0/1, - Master Minimum Equipment List: Doc. no. A-00-18/1-MMLEL Issue/Rev. 0</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/10/2018</td>
<td>Active</td>
<td>10067159</td>
</tr>
<tr>
<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>REV. 1</td>
<td>G 120TP-A</td>
<td>J7689 – Flight Test Instrumentation Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unannounced and remain applicable to this certificate/ approval.</td>
<td>Installation, Operation and Maintenance in accordance with documents control; Master List Document No: Master Data List; GVH J7689-01, MDL-01-(&lt;&gt;)-Issue 4, Dated 01 October 2018&lt;=&gt;(&lt;&gt;), or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 21A.109. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/03/2021</td>
<td>Active</td>
<td>10067160</td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>747-400</td>
<td>Cabin Interior Reconfiguration and IFE System Installation on British Airways Aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unannounced and remain applicable to this certificate/ approval.</td>
<td>Certification Summary, NorthWest Aerospace Technologies, Inc. document No: N0622-D60, Revision A, dated 08 October 2018; Instructions for Continued Airworthiness, NorthWest Aerospace Technologies, Inc. document No: N0622-D70, Revision A, dated 10 September 2018 or later revisions of the above listed doc...</td>
<td>Certification Summary, NorthWest Aerospace Technologies, Inc. document No: N0630-D60, Revision A, dated 10 January 2019; Instructions for Continued Airworthiness, NorthWest Aerospace Technologies, Inc. document No: N0630-D70, Revision A, dated 27 July 2018&lt;=&gt;(&lt;&gt;), or later revisions of the above listed document(s).</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/10/2018</td>
<td>Active</td>
<td>10067166</td>
</tr>
<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>A320-232</td>
<td>Cabin Densification of British Airways Aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unannounced and remain applicable to this certificate/ approval.</td>
<td>Certification Summary, NorthWest Aerospace Technologies, Inc. document No: N0630-D60, Revision A, dated 10 January 2019; Instructions for Continued Airworthiness, NorthWest Aerospace Technologies, Inc. document No: N0630-D70, Revision A, dated 27 July 2018&lt;=&gt;(&lt;&gt;), or later revisions of the above listed document(s).</td>
<td>The following Minor Change Approved Service Bulletins shall be concurrently installed: British Airways SB A320-25-0559, Airbus SBA200-03-1369, Airbus SB A320-25-18GW, Airbus SB A320-53-1596 and Airbus SB A320-25-18HS, Prior to installation of this change/repair it must be determined that the interrelationship...</td>
<td>11/01/2019</td>
<td>Active</td>
<td>10067171</td>
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<tr>
<td>LEE AEROSPACE, INC.</td>
<td>FAA STC SA01867WI</td>
<td>SEE FAA AML</td>
<td>CoolView Compartment Side and Storm Windows Installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>See attached AML No SA01867WI for required data.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/10/2018</td>
<td>Active</td>
<td>10067190</td>
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</table>
10067192
AMAC AEROSPACE SWITZERLAND AG
STC Holder
737-700
Foreign/NAA STC ref.
737-700
Description
Remove Limitation for required cabin crew.
The Certification Basis for the original product remains applicable to this certificate.
Certification basis
The Certification Basis for the original product remains applicable to this certificate.
Associated Technical Documentation
Revision 8.
Limitations/Conditions
The aircraft shall be operated in accordance with AMMC Airplane Flight Manual Supplement AFMS-119-004, Revision 3.
Date of Issue
12/10/2018
Status
Active
System Record
10067192

10067200
VORTEX CONTROL TECHNOLOGIES LLC
FAA STC ST03744NY
737-700
Installation of Vortex Control Technologies Finlets on Boeing 737-700 aircraft.
The Certification Basis for the original product remains applicable to this certificate.
1. ALOFT AeroArchitects Configuration Listing No. 7874-33
CFGM-001 revision C, dated 13 July 2018, or later FAA approved revisions
2. Master Data List No. 7874-001-MDL-(DF+)
Revision 9, dated 11 May 2018, or later FAA approved revisions.
3. Instructions for Continued Airworthiness (ICA), No. TL-877...
Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10067204
LUFTHANSA TECHNIK AG
A321-272N/-279X
Installation of LF-ULD/Lithium Battery.
The Certification Basis for the original product remains applicable to this certificate.
Supplemental Type Design Document - STDC0-18/04 Issue 1 dated 02 OCT. 2018
Compliance Check List - CCL25-18/04M Issue 1 dated 10 OCT. 2018 for later revisions of the above listed document(s) approved/accepted under the EASA system.
Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10067206
AIRBUS HELICOPTERS UK LIMITED
CAA UK RESPONSIBILITY
AS 350 B3
Mod ref. AHUK3503128 consisting of installation and adaptation of Alphagps SAGEM 8531 Thre Axis Autopilot (ref. AW858 1376) to enable IF/Paris Garmin 500i avionics.
The Certification Basis (CB) for the original product remains applicable to this certificate.
MR/3503128-01 "Master Record Design Definition", issue B, dated 9th October 2018
AHUK350-FMS-107 "Flight Manual Supplement", initial issue, dated October 2018
ICA-M3503128-01 "Maintenance / Continued Airworthiness Instructions", issue C, dated 12th October 2018 for later revisions of the above listed document(s).
Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10067211
SABENA TECHNICS BGC
REV. 3
A321-212/213/231-232
GEE Global AirConnect System Activation on Airbus A318/A319/A320/A321/Revision 1: Removal of the Limitation associated with HF
communication system revision 2: introduces the A318/Revision 3: introduces the A320 fitted with AMC2 791 provisions installation of a donut Satellite Communication antenna.
The Certification Basis for the original product remains applicable to this certificate.
- Master Data List, ref 0876-01-A-MDL-R04
  Instructions for Continued Airworthiness (Supplement), ref : 0878-01-A-ICA-R05- AFM (Supplement) 0877-01-A- AFM-R04 later revisions of the above listed document(s) approved/accepted under the EASA system.
The change related to this STC requires as a prerequisite: Aerosonel STC 10067212 or Aerosonel STC 10071558 and Aerosonel/Monor Change 0877-01-A-MCA-R00 or Minor Change 0877-02-A-MCA-R00 and/or Minor Change 0880-01-A-MCA-R00 or later approval revisions.
The installation of the GEE Global AirConnects...

08/11/2019
Active
10067211

### List of EASA Supplemental Type Certificates

<table>
<thead>
<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>10067248</td>
<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td></td>
<td>DR 400 RP, DR 400-120, DR 400-200R, DR 400-900, DR 400-1805, DR 400-212, DR 400-180, DR 400-180R, DR 400-160D, DR 400-140B, DR 400-125, DR 400-125I, DR 400-120A, DR 400-120D, DR 400-100, DR 400 NGL, DR 315, DR 340, DR 360, DR 380, DR 300-140, DR 300-108, DR 250 B, DR 250 B 0-180, DR 220 AB, DR 221, DR 221 B, DR 200, DR 250, DR 250 0-160, DR 200, DR 220 A, DR 220 B, DR 200 AB, DR 221, DR 221 B, DR 200, DR 250, DR 250 0-160, DR 400 RP, DR 400-900, DR 400 NGL, DR 400-212, DR 400-900, DR 400-1805, DR 400-180, DR 400-160D, DR 400-140B, DR 400-125, DR 400-125I, DR 400-120A, DR 400-120D, DR 400-100, DR 400 NGL, DR 315, DR 340, DR 360, DR 380, DR 300-140, DR 300-108, DR 250 B, DR 250 B 0-180, DR 220 AB, DR 221, DR 221 B, DR 200, DR 250, DR 250 0-160</td>
<td>Installation of Avidyne IFD5XX/4XX. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-23, Amdt. 4 -23.1, 29, 301, 305, 307, 603, 605, 609, 611, 771, 773, 853, 1301, 1308, 1308, 1309, 1321, 1327, 1329, 135...</td>
<td>- Description of Modification DE-2053-000000, Issue 2; - Aircraft Flight Manual Supplement PO-2053-000000, Issue 0-1; - Maintenance Manual Supplement MM-2053-000000, Issue 1; - Engineering Work Order EW-2053-000000, Issue 1</td>
<td>10/02/2021</td>
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<tr>
<td>10067249</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td></td>
<td>BD-700-1A10, BD-700-1A11</td>
<td>WLAN Activation Activation of Wireless Local Area Network (WLAN) after physical installation of router including the cockpit WLAN ON/OFF switch with Project AMAC Aerospace Minor Modification EY02-4430.</td>
<td>The change is defined with Revision A of Master Data List Document Number MDL-BD701-4430. The aircraft shall be operated in accordance with the initial issue of Airplane Flight Manual Supplement Document AFMS-BD701-4430 or later revisions of the above listed document(s) approved/accepted under the EASA...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/10/2018</td>
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<td>STC Holders Name</td>
<td>STC holder STC</td>
<td>STC holder STC Revision</td>
<td>Associated Technical Certification Basis</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>FOKKER TECHNIEK B.V.</td>
<td>REV 1</td>
<td>737-800</td>
<td>737-800 from Private use to Commercial Change from Private to both Private</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Airplane Flight Manual Supplement (AFMS) 8738-A004-0601/0 Revision 1: Compliance Record Report CRR-0297 Issue 1 or later revisions of the above listed document(s) approved/accepted/published by the EASA system.</td>
<td>The installation of STC 10047657 is a prerequisite to install this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/12/2019</td>
<td>Active</td>
<td>10047657</td>
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<tr>
<td>JET AVIATION AG</td>
<td>REV 1</td>
<td>FALCON 7X, FALCON 8X (VARIANT)</td>
<td>This STC allows the installation of cargo bags on Dassault Falcon 7X/8X-place divans. The cargo bags are designed for carrying passengerbaggage such as sports equipment. The transport of dangerous goods is excluded.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition (s): ORI D-07-Cargo Seat Bag Installation - The following paragraph(s) at a later amendment: CS-25.25 Amrd. 20 for the paragraphs 25.1, 25.23, 25.301, 25.32.</td>
<td>- Airplane Flight Manual Supplement (AFMS) 8738-M-0104AFM-01/0 Revision 1: Compliance Record Report CRR-0297 Issue 1 or later revisions of the above listed document(s) approved/accepted/published by the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2019</td>
<td>Active</td>
<td>10047659</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>REV 1</td>
<td>FAA STC ST0435NY</td>
<td>Installation of Panasonice TAMDAR.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Elect to Comply paragraph(s) at a later amendment: FAR 23.867 Amrd. 7, FAR 23.1509 Amrd. 672-The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product...</td>
<td>1. Delta Engineering Master Data List 0106-10387-2471 Revision A dated 07 October 2018 or later FAA approved revision 2. The Instructions for Continued Airworthiness as listed on the Master Data List.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the original product. EASA STC 10064143 is required with this STC. Valid for S/N 177RG0252 only. The use of Garmin GNS 430W system is not permitted for instrument approaches using GPS guidance, in accordance with the Limitations specified in AFM-8.</td>
<td>16/11/2018</td>
<td>Active</td>
<td>10047657</td>
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<tr>
<td>DRABPOL SP. JAWANA P. DRABCZYNSKI WISPOLN</td>
<td>REV 1</td>
<td>177 RG</td>
<td>Complete avionic refurbishment as defined in 2-034-014 document.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Findings(s): CR 8-101 - &quot;Vibration buffeting compliance criteria for large external antenna installation at Issue 1The requirements for environmental protection...</td>
<td>1. Delta Engineering Master Data List 0106-10387-2471 Revision A dated 07 October 2018 or later FAA approved revision 2. The Instructions for Continued Airworthiness as listed on the Master Data List.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. EASA STC 10064143 is required with this STC. Valid for S/N 177RG0252 only. The use of Garmin GNS 430W system is not permitted for instrument approaches using GPS guidance, in accordance with the Limitations specified in AFM-8.</td>
<td>19/10/2018</td>
<td>Active</td>
<td>10047659</td>
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<tr>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>REV 1</td>
<td>747-400</td>
<td>Installation of the Antenna and Radar on the top of the aircraft having already received the structural modification approved under initial EASA STC approved revision. The current EASA revised STC approval is covering in addition the SATC...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Findings(s): CR 8-101 - &quot;Vibration buffeting compliance criteria for large external antenna installation at Issue 1The requirements for environmental protection...</td>
<td>Master Manual List, EAD Aerospace document No. 01536-EAS-A-001, Revision D, dated 1st February 2019, and/or Airplane Flight Manual Supplement 01536-EAS-A-001 at revision B dated 14th December 2018, and/or Instructions for Continued Airworthiness, EAD Aerospace document No. 01536-EAS-A-ICA-001, Revision D, and/or Airplane Flight Manual Supplement 01536-EAS-A-ICA-001 at revision B dated 27th May 2022.</td>
<td>See those of Airplane Flight Manual Supplement 01536-EAS-A-AMF-01 revision B, dated 14th December 2018 or later EASA approved revisions and The previous original EASA STC Approval 10062730 was issued on 19th Oct 2018, when the STC configuration definition covered at that time only structural prov...</td>
<td>06/02/2019</td>
<td>Active</td>
<td>10047720</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV 1</td>
<td>FAA STC ST0434NY</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of Adamas Rite infrared Faucet and components Revision 1 add 8737-8 and 8737-4 to the applicability list.</td>
<td>1. Delta Engineering Approved Model List 0198-10338-2473 Revision 0 dated 07 March 2019</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The STC is valid for aircraft types and models a...</td>
<td>04/05/2023</td>
<td>Active</td>
<td>10047721</td>
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<tr>
<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FALCON 2000EX</td>
<td>Installation of a Meggitt IFDO Functional Upgrade (Displays LI5 and Magnetic Heading) in Dassault Aviation Falcon 2000EX aircraft with EASyAvionics installed.</td>
<td>1. Master Data List FOXXVI- D0078-005 Rev IR. dated 30 November 2017, created FAA approved revision.2. Instructions for Continued Airworthiness (ICA) FOXXVI-D0078-155 Rev RIR, dated 28 November 2017, or later FAA approved revision. The ICA and its contents must be incorporated into the operator's maintenance manual...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/10/2018</td>
<td>Active</td>
<td>10087272</td>
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<td></td>
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<tr>
<td>JET AVIATION AG</td>
<td>VVIP Cabin Interior Completion</td>
<td>As per CRI A-4301 The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Airplane Flight Manual Supplement: JBSC-M-00114AFM-01, Issue 02 Certification Compliance Sheet: JBSC-M-00114CSS-01, Issue 05 Master Data List: JBSC-M-00114MDL-01, Issue 03 Instructions for Continued Airworthiness: JBSC-M-00114ICA-01, Issue 01 ICAWS Instructions for Continued Airworthiness: JBSC-M-00114EC...</td>
<td>Limited to S/N 1751 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/10/2018</td>
<td>Active</td>
<td>10087290</td>
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<tr>
<td>CHIPPWA AEROSPACE, Inc.</td>
<td>FALCON 900EX (F900EX)</td>
<td>Installation of an ADS-B Out</td>
<td>Master Drawing List (MDL) No. 170-3402-01-0001, Revision IR, dated 15 January 2016, or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/10/2018</td>
<td>Active</td>
<td>10087298</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FALCON 900EX (F900EX)</td>
<td>Installation of a GoGo AVANCE L5 System in Dassault Aviation Falcon900EX Aircraft</td>
<td>1. Dassault Aircraft Services Master Data List FOXXVI-D0080-005 Rev IR, dated 12 December 2017, or later FAA approved revision.2. Dassault Aircraft Services Instructions for Continued Airworthiness (ICA) FOXXVI-D0080-155, Rev IR, dated 13 December 2017, or later FAA accepted revision.3. Dassault Aircraft Services...</td>
<td>The ICA and its contents must be incorporated into the operator/maintenance program. Prior to activation of a Wireless Cabin Distribution System, it must be ensured that the aircrafts compatible with transmitting portable electronic devices (T-PED) per FAA policy statement PS- ANM25-13: Operational app...</td>
<td>23/10/2018</td>
<td>Active</td>
<td>10087325</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10067334</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC</td>
<td>SA09747DS</td>
<td>F177RG</td>
<td>177B</td>
<td>Installation of an S-TEC 0232 Digital Flight Control System (DFCS).</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval, except where amended by additional or later amendments if indicated on FAA STC SA09747DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certification/approval.</td>
<td>1. Master Data List No. ST-963-MLD-0001, Revision A, dated 30 May 2018 for later FAA approved revision 2. Airplane FlightManual Supplement, ST-963-AMFSS-001, Revision B, dated 30 May 2018</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/10/2018</td>
<td>Active</td>
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<tr>
<td>10067343</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC</td>
<td>SR1123DS</td>
<td>505</td>
<td>Installation of Nightscanner</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval, except where amended by additional or later amendments if indicated on FAA STC SR1123DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certification/approval.</td>
<td>- Master Drawing List, Report No. AA-1708, Revision A, dated 19 July 2018, or later FAA approved revision. - Aeronautical Accessories Instructions for Continued Airworthiness, ST-963-ICA-0001, Revision B, dated 21 May 2020.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/10/2018</td>
<td>Active</td>
<td>10067343</td>
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<tr>
<td>10067345</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>FAA STC</td>
<td>777-300ER/777F</td>
<td>777-200</td>
<td>777-200LR/777-300</td>
<td>ADS-B Upgrade using alternate GPS. The Major Change is required to approve an additional GPS position source for the ADS-B Out transponders. The new GPS position source data is provided by the OMC CMA-5024.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>AFM supplement ref. H44FSM549 Rev. 6, and Modification summary sheet and declaration of compliance ref. H41881 Issue for later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/08/2019</td>
<td>Active</td>
</tr>
<tr>
<td>10067355</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>FAA STC</td>
<td>A330-343</td>
<td>A330-321/322-331/341-342</td>
<td>A330-2431/2431F/301-302</td>
<td>Seat Track and Floor Panels Installation in A330/340/352/362/372/382/392</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>SERVICE BULLETIN S4A-350173-SB001 Edition 1, 18/09/2018 for later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>None.</td>
<td>26/10/2018</td>
<td>Active</td>
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<tr>
<td>10067361</td>
<td>ALAMO ENGINEERING GmbH</td>
<td>FAA STC</td>
<td>EMB-135BJ</td>
<td>EMB-135</td>
<td>EMB-135BJ</td>
<td>Installation of IGA SatCom AntennaRev. 1: Transmitters he STC from “Deutsche Aircraft GmbH” to “AlamoEngineering GmbH”.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>328 Support Services GmbH Master Document List ref. MDL-147201 Issue Dated 1 May 2018 for later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>None.</td>
<td>18/12/2023</td>
<td>Active</td>
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<tr>
<td>10067363</td>
<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>FAA STC</td>
<td>STA0401S</td>
<td>EMB-145EUEMB-145LR/EMB-145MP</td>
<td>EMB-145E/EMB-145LM</td>
<td>Installation of the Honeywell NZ-2000 FMS/GPS/ with or without WAAS/LPV</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval, except where amended by additional or later amendments if indicated on FAA STC STA0401S. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>Chippewa Aerospace Master Drawing List (MDL) No. 145-3451-01-001. Rev. 8, dated 19 December 2018, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/12/2018</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>737-700</td>
<td>MOD AE17150 - G2 Galley Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>- Master DATA LIST Doc: N°: AE17125MDL Rev. 0 dated 29 October 2018- Instructions for Continued Airworthiness AE17110ICA Rev. 0 dated 02/11/2018; later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>- Per STC Approval AE18101IFA/MS - Per Instructions for Continued Airworthiness AE18110ICA - Pre-requisite: Aerotec Change AE16077MAS- This STC constitutes an approval for installation only for aircraft serial numbers as stated in drawing STR-16007-001-000. For additional aircraft serial numbers the require...</td>
<td>02/11/2018</td>
<td>Active</td>
<td>10067414</td>
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<tr>
<td>ETHAD AIRWAYS ENGINEERING L.L.C.</td>
<td>A380-861</td>
<td>EQUIPMENT AND FURNISHINGS - A380 CABIN RECONFIGURATION FROM 11F / 76J / 415 Y TO 11F / 76J / 68 Y + / 337 Y</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>- Master DATA LIST Doc: N°: AE17150MDL Rev. 0 dated 29 October 2018- Instructions for Continued Airworthiness AE17110ICA Rev. 0 dated 31 October 2018; later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>- This design change is applicable only to MSN 166, 170, 176, 185, 191-195, 198, 199, 233 and 237Prior to installation of this change it must be determined the interrelationship between this change and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>03/11/2018</td>
<td>Active</td>
<td>10067398</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>B4-602 R</td>
<td>Installation of a Honeywell Mark V Enhanced Ground Proximity Warning system (EGPWS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>1. EMTEQ Engineering Master Drawing List (MDL) 2-18476/001, Rev. B, dated 30 October 2004, or later FAA approved revision. 2. Flight Manual Supplements (AFMS) for Airbus A300B4-1460 SeriesAircraft, EMTEQdocument No. 2-18476-008, Rev. JR, dated 01 December 2004, or later FAA approved revision.</td>
<td>- Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>02/11/2018</td>
<td>Active</td>
<td>10067419</td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>A300 F4-622 R</td>
<td>Installation of a Honeywell Mark V Enhanced Ground Proximity Warning system (EGPWS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>1. EMTEQ Engineering Master Drawing List (MDL) 2-18476/001, Rev. B, dated 30 October 2004, or later FAA approved revision. 2. Flight Manual Supplements (AFMS) for Airbus A300B4-1460 SeriesAircraft, EMTEQdocument No. 2-18476-008, Rev. JR, dated 01 December 2004, or later FAA approved revision.</td>
<td>- Prior to installation of this design change it must be determined the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>02/11/2018</td>
<td>Active</td>
<td>10067419</td>
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<td>MAXCRAFT AERONAUTICS Ltd.</td>
<td>TCCA STC P-</td>
<td>Approval of Existing Installed AutonocDual GNS430W GPS / NAV / COM, DVC1S-6100 Audio System, KRA-4028 RadarHeightmeter, GTX330 Mode S Transponder, DFS-43A ADF Receiver, KR-21 MarkertBeacon</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) in the latter amendment: Transport Canada AVM 529-3, change 529-3: AVM 529-130-1The requirements for environmental protection and the associated certified...</td>
<td>- Flight Manual Supplement, Garmin GNS430W GPS/NAV/COM (Dual) RMS-05-70-00-091, Revision A dated 26 August 2017- Chapter 04 - Airworthiness Limitations of Instructions for ContinuedAirworthiness, ICA-05-70-00-091, Revision B dated 9 August 2018- Project Development, PQ-05-79-00-091, Revision A dated...</td>
<td>- Approval is limited to S/N 9158 and S/N 9229. This STC does not cover NVIS/NVG operation as well as SBAS based approaches. Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>02/11/2018</td>
<td>Active</td>
<td>10067420</td>
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<td>STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA Code</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10067422</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCDA STC</td>
<td>SH18-9</td>
<td>505</td>
<td>Installation of Dart Aerospace Ltd. Emergency Flotation System in accordance with Master Document List MD655-1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- Master Document List MD655-1 Revision A - Installation Instructions ICA505-1 Revision A - Rotorcraft Flight Manual Supplement RNF8505-1 Revision A - Instructions for Continued Airworthiness ICA505-1 Revision for later revisions of the above listed document(s) approved/accepted/revised of EASA in acco...</td>
<td>Prior to installation of this change/repair it must be determined if the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/11/2018</td>
<td>Active</td>
<td>10067422</td>
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<tr>
<td>10067437</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>REV 1</td>
<td>FAA STC</td>
<td>5021215E</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of Garmin GTS 800 (TAS), GTS 850 (TAS) or GTS 855 (TCAS II) Traffic System.</td>
<td>1. Master Drawing List, GTS 855 Part 21 AM, STC, 005-00738-01, Rev 1, dated 2 May 2014, or later FAA-approved revision. 2. Instructions for Continued Airworthiness ICA505-1 Revision for later revisions of the above listed document(s) approved/accepted/revised of EASA in acco...</td>
<td>Prior to installation of this design change it must be determined if the change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>19/12/2018</td>
<td>Active</td>
<td>10067437</td>
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<tr>
<td>10067443</td>
<td>MORSON PROJECTS LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B200GT, B200CGT</td>
<td>Approval of Category II LLS Approach functionality on BeechKing AirB200GT and B200CGT in accordance with Morson Projects Modification refMP0005A.</td>
<td>The Certification Basis for the original product and its amendments (including the following additional or alternative airworthiness requirements: CS 23A Amendement 4) for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Morson Projects Modification refMP0005A and Service bulletin ref SB MP34-007 latest FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined if the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/11/2018</td>
<td>Active</td>
<td>10067443</td>
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<tr>
<td>10067445</td>
<td>ETHAD AIRWAYS ENGINEERING L.L.C.</td>
<td></td>
<td></td>
<td>787-9</td>
<td>REPLACEMENT OF ELECTRICAL GALLEY INSERT.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Master Document List ref EB8-1652784-02 13, issued 9th July 2020, or later FAA-approved revision.</td>
<td>Prior to installation of this change/repair it must be determined if the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/11/2018</td>
<td>Active</td>
<td>10067445</td>
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<tr>
<td>10067448</td>
<td>ALOFT AEROCRAFTI T S</td>
<td>REV. 3</td>
<td>FAA STC</td>
<td>ST04052NY</td>
<td>ViSat-A KA Cabin Distribution System Installation Rev. 2: Applicability list extended to B737-8/9R-3: Amendment to add configuration 3 onto Boeing 737-8 aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>1. Master Data Index List (MDIL) 12007351, revision 013, dated 27 June 2023, or later FAA-approved revision. 2. Instructions for Continued Airworthiness ICA505-1 Revision for later revisions of the above listed document(s) approved/accepted/revised of EASA in acc...</td>
<td>The following alterations are prerequisites to this installation: EASA STC 10058725 Installation of a ViSat-A KA Satellite Antenna System. Prior to activation of a Wireless Cabin Distribution System, it must be ensured that the aircraft is compatible with transmitting portable electronic devices (T-PED) per F...</td>
<td>26/10/2023</td>
<td>Active</td>
<td>10067448</td>
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<td>10067464</td>
<td>C&amp;D ZODIAC, Inc.</td>
<td>FAA STC</td>
<td>ST03961A</td>
<td>ERU 190-100 LR</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if included on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>C(1&lt;-6)=C(1&lt;-9)=D Zodiac, Inc. Interior Installations and Modifications.</td>
<td>Prior to installation of this index List No. 0811, Revision A, dated 10 November 2014, or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined if the change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC approves provisions for the installation...</td>
<td>07/11/2018</td>
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<td>10067464</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>AEROLITE AG</td>
<td>REV. 4</td>
<td>PC-24</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>General Installation according to Aerolite Master Document List MDL-76-005, Rev.M. For EMS Equipment PC-24/PN 024-0229560-S01: Aerolite Flight Manual Supplement FMS-024-0229562-S01, Rev. 3: Installed in accordance with the Appendix 1 of the Flight Manual Supplement Doc. No. 5G-1027-FMS-01 and the Appendix 1 of the Cabin Crew Data Supplement Doc. No. 5G-1027-CCD-01 to include the appropriate cabin interior lighting configurations and &quot;Charge...</td>
<td>12/05/2023</td>
<td>Active</td>
<td>10067483</td>
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<td>GAL AVIATION Inc.</td>
<td>TOCA STC:</td>
<td>DHC-8-402</td>
<td>Fabrication and Installation of New Overhead Bins</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>- The modification and installation must be carried out in accordance with Can-Am Aerospace Master Drawing List 25-17-08-005, Revision NC dated 7th of March 2018. The maintenance must be carried out in accordance with Can-Am/Aerospace Instructions for Continued Airworthiness (CA) 25-17-08-00RevA...</td>
<td></td>
<td>12/11/2018</td>
<td>Active</td>
<td>10067500</td>
<td></td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>FAA STC</td>
<td>SAAB S340A, SAAB 340B</td>
<td>Replacement of certain Goodrich pneumatic de-ice boots</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>SMR Technologies, Inc Pneumatic Device/Part Layout Document No.SMR54141, Revision 8, dated 8 October 2017 and detail design as listed on MDL-0052, Revision N, dated 18 May 2018 or later FAA approved revision.</td>
<td></td>
<td>14/11/2018</td>
<td>Active</td>
<td>10067510</td>
<td></td>
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<tr>
<td>AERODATA AG</td>
<td>B300</td>
<td></td>
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<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td>MMEL: OSD-MMEL-1000 Rev. 1, dated 2 November 2018: FCD: OSD-FCD-1000 Rev. 0, dated 25 September 2018 or later revisions of the above listed document(s) approved/acceptable the EASA system</td>
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<td>14/11/2018</td>
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<td>10067510</td>
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<tr>
<td>LORD</td>
<td>Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>Aircraft Flight Manual Supplement AFM-165 Rev. 1 dated 26-Oct-2018, Aircraft Maintenance Manual Supplement AMM-1168 Master Document List No. MDL3815002 Revision dated 15 October 2018, or later revisions of the above listed document(s) approved and accepted.</td>
<td>Limited to B300 aircraft equipped with Rockwell Collins Proline Fusion Avionics. Prior to installation of this change/repair it must be determined whether the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/11/2018</td>
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<td>10067519</td>
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<td>AIRBUS</td>
<td>Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>Aircraft Flight Manual Supplement AFM-165 Rev. 1 dated 26-Oct-2018, Aircraft Maintenance Manual Supplement AMM-1168 Master Document List No. MDL3815002 Revision dated 15 October 2018, or later revisions of the above listed document(s) approved and accepted.</td>
<td>Prior to installation of this change/repair it must be determined whether the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/11/2018</td>
<td>Active</td>
<td>10067569</td>
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<td>SPECIALIST</td>
<td>Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>- Certification Summary Report ref. A319/160673 Issue 6, dated 29-Nov-2017 - Modification Approval Sheet ref. MAS A40074-6 Issue 2, dated 29-Nov-2018.</td>
<td>None</td>
<td>21/11/2018</td>
<td>Active</td>
<td>10067569</td>
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<td>AIRBUS</td>
<td>Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and emissions levels of the original product unaltered and remain applicable to this certificate approval.</td>
<td>Aircraft Flight Manual Supplement AFM-165 Rev. 1 dated 26-Oct-2018, Aircraft Maintenance Manual Supplement AMM-1168 Master Document List No. MDL3815002 Revision dated 15 October 2018, or later revisions of the above listed document(s) approved and accepted.</td>
<td>Prior to installation of this change/repair it must be determined whether the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/11/2018</td>
<td>Active</td>
<td>10067626</td>
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# List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
<th>STC Holder</th>
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<th>Foreign/NAA STC ref.</th>
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<tr>
<td>10067682</td>
<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>737-800</td>
<td>Kontron CabinConnect Wireless IFE System. The modification installs the Kontron CabinConnect Wireless IFE System on the Boeing 737-800 aircraft. The Cabin Wireless IFE system provides seamless access to audio-visual infl-entertainment content to passengers via carry-on PEDs...</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (a): SC H-21. ICA on EWI. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product...</td>
<td>- Master data list report TEN 840 issue 1; Airplane flight manual supplement report TEN 840-47 issue 1. - Instructions for Continued Airworthiness Report TEN 840-04 issue 2; or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/11/2018</td>
<td>Active</td>
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<td>10067685</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST01824W</td>
<td>MYSTERE FALCON 900 (MF900)</td>
<td>Installation of a Data Link Communication System supporting future AirNavigation (FANS IIA) capability.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (c): EASA CRF 9122-002, EASA CRF-9122-003. The requirements for environmental protection and the associated certified noise and/or emissions levels of the above listed document(s)...</td>
<td>EDOC-020049 Rev C Master Document List - March 24, 2017; EDOC-020085 Rev C AFMS - March 24, 2017, or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>None.</td>
<td>20/12/2018</td>
<td>Active</td>
<td>10067685</td>
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<tr>
<td>10067699</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>A330-200, A330-300, A330-343</td>
<td>Large KA-band antennas and systems Installation – A330 Family Glnow collectively Rev 1. Extension of STC 10067699 applicability to A330-343. Model: Design optimization affecting Structural &amp; F&lt;(&gt;&amp;&lt;) OT Analysis foundation models.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Equivalence Safety Finding(s): CRI B-101 Vibration buffeting compliancerequirement for large antenna installation issue 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the above listed document(s)...</td>
<td>- Supplemental Type Design Document STD203-17/11M Issue 2 Rev 0; Supplemental Type Design Document STD203-17/11M Issue 3 Rev 0 dated 21 April 2020.</td>
<td>Installation allowed only on models that satisfy, respectively, the criteria and configuration defined in the configuration sheet AWPX01-23-F2003-HDK02 Service bulletin Edt F2003-HDK02 Rev 2 dated 21 Nov. 2018 and AWPX01-53-E3-002 Rev 3 dated 21 Nov. 2018, or later revisions of these documents. Prior to installation of this change/repair...</td>
<td>06/05/2020</td>
<td>Active</td>
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<tr>
<td>10067717</td>
<td>SAA SOLUTIONS FOR AVIATION, S.L</td>
<td>REV. 2</td>
<td>MYSTERE-FALCON 200</td>
<td>INSTALLATION OF FOR LI MODAL F1000 IN FALCON 20 AIRPLANE</td>
<td>The Certification Basis for the original product remains applicable if the certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>- Supplemental Type Design Document STD203-17/11M Issue 2 Rev 0; Supplemental Type Design Document STD203-17/11M Issue 3 Rev 0 dated 21 April 2020.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/12/2018</td>
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<td>10067717</td>
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<tr>
<td>10067723</td>
<td>AGUSTA WESTLAND PHIADELPHIA PORATION</td>
<td>FAA STC SR04230NY</td>
<td>AW119MKII</td>
<td>Installation of a TRAKOA A860 Searchlight</td>
<td>The Certification Basis for the original product remains applicable if the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>1. AgustaWestland Philadelphia Corporation Master Data List FA-05-02003 ENG-17-061 Revisions, dated 20 June 2018 or later FAA approved revisions.2 Maintenance must be in accordance with AgustaWestland Philadelphia Corporation Instructions for Continued Airworthiness, for the TRAKOA860 Searchlight.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/12/2018</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>REV. 2</td>
<td>FAA STC ST02618A</td>
<td>747-400, 747-400F</td>
<td>Upgrade of the Aviation Communication &amp; Surveillance Systems (ACSS)X5-950 Mode S ADS-B Out Transponder or Installation of the ACSS XKT-800 Mode S/ADS-B Out Transponder including ADS-B Out functionality. Rev 3 modifies a Standalone GPS source to provide Navigation 9114a (to the ACSS X5-950 or ACSS NX...</td>
<td>ACSS ST 965 of Master Data List, Drawing No. 8517447, Revision 1, dated 29 September 2018, or later FAA approved revision, for Continued Airworthiness (ICA) and FAA approved Airplane Flight Manual Supplement (AFMS), called out into corresponding FAA approved Master Data List (MDL) listed...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2019</td>
<td>Active</td>
<td>1006774C</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SA09749DS</td>
<td>320E, 320F, 335, 340, 340A, 310A, T310R, 320B, 320C, 320D, 310H, T310P, 310Q, T310G, 310Q, 310J, E310J, 310K, 310L, 310D, 310E, 310F, 310G, 310H</td>
<td>Installation of an S-TEC 01326 Digital Flight Control System (DFCS)</td>
<td>Master Data List No. ST-965-AML-0001, Revision A, dated 29 August 2010 for later FAA approved revision, Airplane Flight Manual Supplement, ST-965-AFMS-0001, Revision 1, dated 30 September 2018 or later FAA approved revision, for Continued Airworthiness (ICA) and FAA approved Airplane Flight Manual Supplement, ST-965-AML-0001, Revision A, dated 29 August 2010 for later FAA approved revision, for Continued Airworthiness (ICA).</td>
<td>Prior to installation of this design change must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/05/2018</td>
<td>Active</td>
<td>1006775D</td>
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<tr>
<td>SWISS AIR-AMBULANCE Ltd.</td>
<td></td>
<td>AW109SP</td>
<td></td>
<td>Replacement of existing incandescent lamps with LED lamps for landing, taxiing- and cabin searchlight (ref. modification RE.10344) for NVIS certified AW109SP helicopter</td>
<td>Master Drawing List RE. 10344.DW, issue 1, dated 29 September 2018. ICA Supplement, RE. 10344.ICS, issue 2, dated 07 November 2018. Service Bulletin RE. 10344.SB.01, issue 03, dated 06 November 2018. Certification report, RE. 10344.CR, issue 01, dated 06 September 2018.</td>
<td>Prior to installation of this design change must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/12/2018</td>
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<td>AHCON INDUSTRIES, Inc.</td>
<td>REV. 5</td>
<td>FAA STC SA2429CE</td>
<td>747-400, 747-400F</td>
<td>Installation of the S-TEC Flight Manual Supplement (AFMS), called out into corresponding FAA approved Master Data List (MDL) listed...</td>
<td>Avcon Drawing List No. 1043, dated 1 November 1988, or later FAA approved revision, see FAA Approved Model List (AML) No. SA2429CE for PA31's Operating Handbook and FAA Approved Airplane Flight Manual Supplements.</td>
<td>These airplanes are in a Restricted Category when this modification is installed. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2020</td>
<td>Active</td>
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<td>WENDLING, TOBIAS</td>
<td>ASW 20, ASW 20 L</td>
<td></td>
<td></td>
<td>Installation of winglets 'Maughner' kind of instead of winglets 'Lindner' kind of.</td>
<td>Dimension of winglets defined in Tobias Wendling document 'AbmaßeWinglets', at revision 1.1, dated 11 November 2018. Installation in accordance with Tobias Wendling document 'Beschreibung', at revision 1.1, dated November 13, 2018, Airplane Flight and Maintenance Manual Supplement, Tobias Wendling...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/12/2018</td>
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<td>EASA STC no.</td>
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<td>Associated Technical Documentation</td>
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<td>10067829</td>
<td>S-TEC CORPORATION</td>
<td>FAA STD SA09748D5</td>
<td>PA-32RT-300T (TURBO LANCE II)</td>
<td>Installation of a S-TEC 01326 Digital Flight Control System (DFCS). The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original STC...</td>
<td>Master Data List ST-957-4DL-0001, Revision A, dated 2 July 2018 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/12/2018</td>
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<td>10067840</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>A330-243</td>
<td>Crew rest installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. Except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original STC...</td>
<td>Air France Service Bulletin (Service Bulletin issued by AirFrance DOA) AF284330170102S-001, Air France DOA Change Approval Sheet CAO-A330-17-010, or later revisions of the above listed document(s) approved/acceptable to the EASA.</td>
<td>No adverse effect upon the airworthiness of the product.</td>
<td>06/12/2018</td>
<td>Active</td>
<td>10067840</td>
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<tr>
<td>10067844</td>
<td>GLOBAL AEROSPACE DESIGNCORPORATION</td>
<td>FAA STC ST04321CH</td>
<td>737-800</td>
<td>Install passenger cabin upgrade. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Global Aerospace Design Corp Master Drawing List GR0362-MDL, Revision A, dated 4 May 2018, or later FAA approved revisions.</td>
<td>No adverse effect upon the airworthiness of the product.</td>
<td>06/12/2018</td>
<td>Active</td>
<td>10067844</td>
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<td>10067883</td>
<td>FTI ENGINEERING NETWORK GmbH</td>
<td>EMB-145MP, EMB-145SU, EMB-145MK, EMB-145EU, EMB-145LR, EMB-145SEP, EMB-145ER, EMB-155LR, EMB-145, EMB-155BJ, EMB-15SER</td>
<td>A350-B Out for EMB-145 Series: Note: This STC has been updated to Rev. 1 for the definition of an alternate GPS Antenna Position. The Certification Basis for the original product remains applicable to this certificate/approval. Except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- FTI Master Document List document ref. F18-EO15-1634 Issue 4 dated 10 December 2019; FTI AFM Supplement documents ref. F20-001-EO18-1634 Issue 4 dated 22 October 2019 and F20002-EO18-1634 Issue 3 dated 11 December 2018; FTI Instructions for Continued Airworthiness document ref. F33-EO18-1634 Issue 5.</td>
<td>The Airworthiness Limitations as defined in chapter 2 of FTI ICADocument ref. F33EO18-1634 are applicable to this STC installation.</td>
<td>12/12/2019</td>
<td>Active</td>
<td>10067883</td>
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<td>10067889</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST02218AT-D</td>
<td>767-200, 767-400</td>
<td>Installation of Collins HF Data Link System. The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original STC...</td>
<td>TIMCO Master Data List 98T27905, Revision G, dated 30 June 2001, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/12/2018</td>
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<td>Associated Technical Documentation</td>
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<td>10067871</td>
<td>AEROSPACE DESIGN &amp; COMPLIANCE LLC</td>
<td>FAA STC</td>
<td>ST04010NY</td>
<td>A321-231N / -232N</td>
<td>Installation of Routed Polystyrene Foam (RPF) for Center Fuel Tank Ignition Mitigation</td>
<td>1. Aerospace Design &amp; Compliance, LLC. Master Data List A321- MDL-0004 Rev RR, dated 23 April 2018, or later FAA approved revision 2. Instructions for Continued Airworthiness (ICA) A321- ICA-0004, Rev RR, dated 16 March 2018, or later FAA accepted revision 3. Aircraft Flight Manual Supplement (AFMS) A321-</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce harm to the airworthiness of the product.</td>
<td>07/12/2018</td>
<td>Active</td>
<td>10067871</td>
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<tr>
<td>10067872</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>FAA STC</td>
<td>ST01811WI</td>
<td>DC-9-32</td>
<td>Installation of a New SISFDR L3 Model FA2100, and U/L DUKANE DK-12090</td>
<td>1. Boeing MD-82 MSN 49902, Boeing MD-83 MSN 49906. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce adverse effect upon the airworthiness of the product.</td>
<td>07/12/2018</td>
<td>Active</td>
<td>10067872</td>
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<tr>
<td>10067876</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC</td>
<td>ST01736WI</td>
<td>MYSTERE FALCON 900 (MF900)</td>
<td>Installation of a Cobham Aviator 7000 Satellite Communication System (SATCOM)</td>
<td>EDOC-019881 Rev B MDL Master Data List - Sept, 24, 2015;EDOC-018689 Rev B Instructions for Continued Airworthiness - Sept, 24, 2015;EDOC-018700 Rev A Airplane Flight Manual Supplement - June 03, 2015;or later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Applicable to MF900, A and B models. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce adverse effect upon the airworthiness of the product.</td>
<td>07/12/2018</td>
<td>Active</td>
<td>10067876</td>
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<tr>
<td>10067947</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC</td>
<td>ST01811WI</td>
<td>HAWKER 800, HAWKER 800XP</td>
<td>Honeywell Primus II ADS-B Out The Honeywell Primus II Systems upgraded to provide ADS-B Out capability. This includes upgrading the existing Honeywell Integrated Control Unit and the GPS/INS and installing a new Honeywell RGS 200 GPS/INS as position source.</td>
<td>Installation in accordance with Duncan Aviation Inc. MasterDocumentList EDOC-019890, Rev. B, dated 8th October 2018;Operation in accordance with Duncan Aviation Inc. Aircraft Flight Manual Supplement EDOC-018986, Rev. A, dated 4th April 2016;Maintenance in accordance with Duncan Aviation Inc. Instru...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce adverse effect upon the airworthiness of the product.</td>
<td>10/12/2018</td>
<td>Active</td>
<td>10067947</td>
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<tr>
<td>10067948</td>
<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>ST00368O</td>
<td>CL-600-2B16 (604) VARIANT</td>
<td>The STC installs a Honeywell Jetwave™ KA Band Satellite Data System.</td>
<td>1. Bombardier Master Document List (MDL), Document number BA363017019,Revision C, dated 1 May 2017, or later FAA approved revision 2. Learjet Inc.'s Aircraft Flight Manual Supplement (AFMS) DocumentNumber BA363047020, revision NC, dated and FAA approved on 3 May 2017 or later FAA approved revisions...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce adverse effect upon the airworthiness of the product. The aircraft must have the fin top radome M38-10...</td>
<td>18/12/2018</td>
<td>Active</td>
<td>10067948</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Rev.</td>
<td>Foreign/NAA STC No.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10067952</td>
<td>CARLISLE INTERCONNECT TECHNOLOGIES</td>
<td>FAA STD ST0424CH</td>
<td>777-300, 777-300ER</td>
<td>Installation of the Outside Antenna Equipment Provisions for the RINC791 SATCOM System.</td>
<td>The Certification Basis for the original product as amended shall include all additional or alternative aircraft noise reduction requirements.</td>
<td>Equivalent Safety Finding(s): CIR B-101 “Vibration/ buffing compliance criteria for large external”. The requirements for environmental protection and the associated certificate/approval.</td>
<td>- Master Data List (MDL), Ref. 900-429653 Rev. E, Aircraft Flight Manual Supplement (AFMS), Ref. 900-431272 Rev. N, Instructions for Continued Airworthiness (ICA), Ref. 900-394951 Rev. B. For later revisions of the above listed document(s) approved/accepted, one shall title of EASA in accordance with the Tec...</td>
<td>Per Section 4 of the Instructions for Continued Airworthiness (ICA). As per AFMS DC303-1002-00 Revision B. In particular, the EUT does not need to be manually activated for test purposes. Manual activation, if desired, is limited to the first five minutes of the hour and for the point in time and for a period of time in which the EUT is not in use.</td>
<td>11/12/2018</td>
<td>Active</td>
</tr>
<tr>
<td>10067962</td>
<td>ARMSTRONG AEROSPACE Inc d/b/a AERONAUTICS ARMSTRONG AEROSPACE</td>
<td>FAA ST01918CH</td>
<td>A320-941, A320-341-1240-343, A320-321-322-323, A320-331-322-323, A330-301-302-303, A330-341-345 F, A330-223-223F, A330-202-222-243, A330-201-202-203</td>
<td>Installation of Honeywell RESCU406AF Iridium Satellite Communications System Installation.</td>
<td>The Certification Basis for the original product as amended shall include all additional or alternative aircraft noise reduction requirements. Special Condition (s): SC F-11 Non-rechargeable Lithium Battery Installation. The following paragraph(s) at a later amendment: CIS 25-25a, (a,b) amended 19CS 25-301 (a,b) amended 19CS 25.</td>
<td>MDL DC104-1002-00 Rev. R dated March 11 2020 AFAMS DC303-1002-00 Rev. B dated October 20 2009/CA DC113-1002-00 Rev. F dated August 28 2009/DC113-1009-00 Revision A dated February 10 2004/DC11-880746-00 Revision E dated March 10 2002/030 later revisions of the above listed document(s) approved/accepted...</td>
<td>As per AFMS DC303-1002-00 Revision B. In particular, the ELT does not need to be manually activated for test purposes. Manual activation, if desired, is limited to the first five minutes of the hour and for the point in time and for a period of time in which the ELT is not in use.</td>
<td>31/08/2020</td>
<td>Active</td>
<td>10067962</td>
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<tr>
<td>10067971</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>FAA ST/137</td>
<td>A320-214</td>
<td>Ka-band Satcom System Installation Revision 1: ICA ASL structural inspection intervals change.</td>
<td>The Certification Basis for the original product as amended shall include all additional or alternative aircraft noise reduction requirements.</td>
<td>Supplemental Type Design Document, Ref. STD053-18086 Issue02 Revision00 Aircraft Flight Manual Supplement, Ref. A-00-1866-4AFM Issue 01 Revision00 Flight Crew Data, Ref. A-00-1866-FCD Issue 02 Instruction for Continued Airworthiness, Ref. A-00-1866/ICA Issue 02 Revision 00 later rev...</td>
<td>As per: Aircraft Flight Manual Supplement, Ref. A-00-1866-4AFM-Instructions for Continued Airworthiness, Ref. A-00-1866-5AFM-ICA. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/07/2019</td>
<td>Active</td>
<td>10067971</td>
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<tr>
<td>10067962</td>
<td>EMB TECHNOLOGIES CANADA, Ltd.</td>
<td>FAA ST/137</td>
<td>429</td>
<td>Aspire 200 IG SATCOM with Cabin Wi-Fi System Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Avionics Design Services Ltd. Master Drawing List MLD17032/Revision/WC, Transport Canada approved September 29, 2017/EMB Technologies Flight Manual Supplement FMS17032-1 Revision/WC, Transport Canada approved September 29, 2017/EMB Technologies Instructions for Continued Airworthiness ICA17032-1 Re...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/12/2018</td>
<td>Active</td>
<td>10067962</td>
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<tr>
<td>10067995</td>
<td>WEST STAR AVIATION, LLC/ a/k/a WEST STAR AVIATION</td>
<td>FAA STC ST00914DE</td>
<td>MISTERE FALCON 900 (MF900)</td>
<td>Installation of an Iridium Satellite Communications Data Link Unit.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments.</td>
<td>- Master Data List Ref. 18311-17-1751, Revision B, dated 10 October 2018, or later approved FAA revision 2: Aircraft Flight Manual Supplement, West Star Aviation. Doc. No. 18311A-1752, Revision II, R dated 31 July 2018, FAA approved 8 October 2018 or later FAA approved revisions.3. Instructions for Continued Air...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/12/2018</td>
<td>Active</td>
<td>10067995</td>
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<td>STC Holder</td>
<td>FAA Part 150 STC</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>STC Holder</td>
<td>Upgrade of Honeywell Primus II System</td>
<td>737-8 8 MAX, 190Y ACRO Series 6</td>
<td>Installation, Change in ACRO Series 6 seats in a 190Y passenger configuration into a 8 MAX configuration from the TC holder.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Document LIST MLO-1087-01</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/12/2018</td>
<td>Active</td>
<td>10066002</td>
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</tr>
<tr>
<td>Duncan Aviation, Inc.</td>
<td>FAA ST0810W</td>
<td>560 AND 560XL</td>
<td>Upgrade of Honeywell Primus II System for ADS-B OUT with existing/universal UNIC-Escape Position Source or installation of stand-alone Honeywell KGS-200 position source.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Document LIST MOD-1087-01</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/12/2018</td>
<td>Active</td>
<td>10066010</td>
<td></td>
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<tr>
<td>Comtech Global Aviation Services</td>
<td>FAA Part 150 STC</td>
<td>A320-231/-232/-233</td>
<td>Introduction of Centre Fuel Tank Ignition Mitigation Reactivated Polyurethane Foam.</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: 25.303; 25.321(a); 25.303; 25.307(a); 25.332; 25.333; 25.401; 25.603; 25.606(a); 25.609; 25.611; 25.901(1)</td>
<td>Master Document LIST MOD-1087-01</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/04/2019</td>
<td>Active</td>
<td>10068013</td>
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<tr>
<td>Dassault Falcon Jet</td>
<td>FAA ST04293</td>
<td>MYSTERE FALCON 900</td>
<td>Battery Installation (36 or 40 hour) in accordance with Drawing List Report No. F90-01053, Revision G, dated 18 October 2001, or later FAA approved revision.</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: 25.303; 25.321(a); 25.303; 25.307(a); 25.332; 25.333; 25.401; 25.603; 25.606(a); 25.609; 25.611; 25.901(1)</td>
<td>Master Document LIST MOD-1087-01</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2018</td>
<td>Active</td>
<td>10066020</td>
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<tr>
<td>Standard Aeronautical Services ES, LLC.</td>
<td>FAA ST04293</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Installation of a Honeywell JetWave System.</td>
<td>Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC STA23259O-D. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document LIST MOD-1087-01</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2018</td>
<td>Active</td>
<td>10068025</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NA A/C ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10068074</td>
<td>TENENGA Ltd</td>
<td>REV. 1</td>
<td>CA-120-6444-REV. 1</td>
<td>737-9, 737-800, 737-900ER 737-700, 737-800, 737-8</td>
<td>TEN 845Installation of an auxiliary EAI-1V that provides provisions for RUs to be installed in the Electronics and Engineering Bay. The EAI-1Vs located between F3 312 - F3 326 and LBL 3.75 - LBL 27.4 at WL1200. Its weight limited to hold up to 50 lbs of equipment.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List EASA STC 10068074 issue 1</td>
<td>07/12/2019</td>
<td>Active</td>
<td>10068074</td>
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<tr>
<td>10068090</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>A321-213/-211/-232</td>
<td>A321-231/-232/-233/-251N</td>
<td>A320-211/-212/-214/-215/-216 A319-131/-132/-133 A320-111/-121/-114/-115</td>
<td>Wireless (WFi) Infotainment System/This modification activates a Wireless (WFi) Infotainment (Cabin/Distribution) System. The revision allows for the eligibility of the A320-251N (A320neo) for existing STC 10068090.</td>
<td>- Compliance Check List – CCL44-18/08M Rev 4 dated 19 MAR 2019 CCL44-18/05M Rev 1 dated 04-Jun-2019.</td>
<td>- Aviation Flight Manual – A-00-18/103-4FM Issue 01 Rev 01 dated 03-May-2019 - Instruction of Continued Airworthiness – A-00-18/103-ICA Issue 02 Rev01 dated...</td>
<td>- - Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/06/2019</td>
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<tr>
<td>10068097</td>
<td>DART AEROSPACE Ltd</td>
<td>REV. 1</td>
<td>TCCA STC SH18-49</td>
<td>EC135 T2+ EC135 T2 EC135 T1 EC135 P2+ EC135 P2 EC135 P1</td>
<td>Installation of Dart Aerospace Ltd. onsets in accordance with Master Document List MDL-D135-1027 with reference to TCCA STC SH18-49.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: CS 27 Amendment 4 for the following paragraphs: 27.1; 27.2; 27.21; 27.25; 27.27; 27.29; 27.29.7; 27.23; 27.241; 27.251; 27.30...</td>
<td>- Amends the Compliance Report ACR-D135-1027-1 Revision A Master Document List MCDL-D135-1027 Revision B Installation Instructions IN-D135-1027 C-Instructions for Continued Airworthiness ICA-D135-1027 Revision for later revisions of the above listed document(s) approved/accepted on...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/12/2018</td>
<td>Active</td>
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<tr>
<td>10068104</td>
<td>STG AEROSPACE GERMANY GmbH</td>
<td>REV. 1</td>
<td>757-200-300 737-600-700-800-900-900ER 737-100-200-300-400-500</td>
<td></td>
<td>REPLACEMENT PSU LENS ASSEMBLY Revision 1: Transfer of STC holder from STG Aerospace Limited to STGAerospace GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>STG AEROSPACE LTD AEROPLANE FLIGHT MANUAL SUPPLEMENT No. SG-1016-FMS-01/Revision IRSTG AEROSPACE LTD INSTRUCTIONS FOR CONTINUED AIRWORTHINESS No. SG-1016-ICA-01 Revision IRSTG AEROSPACE LTD MASTER DATA LIST No. SG-1016-MDL-01 Revision IRSTG AEROSPACE LTD for later revisions of the above listed document(s) approved/accepted...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2021</td>
<td>Active</td>
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<tr>
<td>10068135</td>
<td>CENTEVRIL MAINTENANCE SL</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of MDU001-03 recording unit for operational purposes.</td>
<td></td>
<td></td>
<td>For applicability of the following documents refer to EASA approved ANLM/001-2018-02 “Master Drawing List”, Revisions dated 21 November 2018FMS-02/2018-03 “Flight Manual Supplement”, issue 01 Rev 00, 10 July 2018ICA-2012-1 “Instructions of Continued Airworthiness”, Issue 01 Rev 02 dated 07 June 2019ICA-2010...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/12/2018</td>
<td>Active</td>
<td>10068135</td>
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<td>EASA STC no.</td>
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<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10068155</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 2</td>
<td>CAAR-UN</td>
<td>STC Holder</td>
<td>Installation of Collins IDC-900. System Modification of an existing FAA approved Active CAA UK Active</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10068156</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>FAA STC</td>
<td>MOD AE17144 : AEROSAT F-310 External Antenna (KU BAND) and System Installation/Revision 1: Correction of pre-requisite modification AE17026 instead of AE17126</td>
<td>Certification basis</td>
<td>- Master Data List document N. &quot;AE17144MDL Revision 0 dated 17 December 2018: Instructions for Continued Airworthiness document N. AE17144MDLRevision 0 dated 07 December 2018;or later revisions of the above listed document(s) approved acceptably under the EASA system.</td>
<td>Applicable to CST/CPS cockpit variants only,A fully functional and approved installation of Garmin G7100/G750, as well as a suitable XIFD100-Antenna must have been already installed. Prior to installation of this change/repair it must be determined thatthe integration line between this change/repair...</td>
<td>08/01/2019</td>
<td>Active</td>
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<tr>
<td>10068157</td>
<td>ADAC LUFTFAHRT TECHNIK GmbH</td>
<td></td>
<td>FAA STC</td>
<td>ST01520WI</td>
<td>Installation of ADS-B IN/OUT System in accordance with Master Document List EC135-08679-001-LMD.</td>
<td>Certification basis</td>
<td>- Rotorcraft Flight Manual Supplement EC135-08769-025-MFS Revision 0: Master Document List EC135-08679-001-LMD Revision 0: Compliance Summary EC135-08679-007-RCS Revision 2: Maintenance Manual Supplement ICA EC135-08679-033-MMS Revision(s) or later revisions of the above listed document(s) approved.</td>
<td>Applicable to CST/CPS cockpit variants only,A fully functional and approved installation of Garmin G7100/G750, as well as a suitable XIFD100-Antenna must have been already installed. Prior to installation of this change/repair it must be determined thatthe integration line between this change/repair...</td>
<td>20/12/2018</td>
<td>Active</td>
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<tr>
<td>10068159</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td></td>
<td>FAA STC</td>
<td>ST01520WI</td>
<td>Installation of the Concorde RG-442 Battery and Temperature Sensorin place of the original Nickel-Cadmium Battery.</td>
<td>Certification basis</td>
<td>Master Drawing List, Drawing No. S-0278, Revision NC, dated 05 June 2008,or later FAA approved revision.</td>
<td>Prior to installation of this change design it must be determined thatthe interaction line between this change design change and any other previsouly installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/12/2018</td>
<td>Active</td>
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<td>10068183</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td></td>
<td>FAA STC</td>
<td>ST01185WI-D</td>
<td>Modification of an existing FAA approved Collins HF Communication System with/ HF-400D Receiver/Transmitter, CPN 822-0990-002/003, and CPL-820DHF Antenna Coupler. CPN 822-0987-003, to a Collins HF Communication System with a HF-400D Receiver/Transmitter, CPN 822-0990-004, and CPL-820DHF Antenna.</td>
<td>Certification basis</td>
<td>Collins Index as listed on FAA STC ST01185WI-D.</td>
<td>Prior to installation of this change design it must be determined thatthere interaction line between this change design change and any other previsouly installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/12/2018</td>
<td>Active</td>
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<tr>
<td>10068184</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td></td>
<td>FAA STC</td>
<td>ST0217AT-D</td>
<td>Modification of Collins IDC-900.</td>
<td>Certification basis</td>
<td>TIMCO Master Drawing List 5BT0126L004, Revision E, dated 14 August 2001,or later FAA approved revision.</td>
<td>Prior to installation of this change design it must be determined thatthere interaction line between this change design change and any other previsouly installed design change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/12/2018</td>
<td>Active</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>10068185</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>-</td>
<td>M7770B ADS-B Upgrade This 4ML STC consists of replacement of existing dual M7677A XFPOR unit by dual M770B XFPOR (compliant to European 2020 Mandate) and installation of an ADS-B 1FAIL/ADS-B 2 FAIL Annunciator or, alternatively, the Gables G7584 Control panel may be used for ADS-B failure indication.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 2 and CS ACN 6 Subpart D. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are...</td>
<td>Aircraft Flight Manual Supplement, Doc Ref: 2111229-AM Instructions for Continued Airworthiness, Doc Ref: 2111229-AM Lateral revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>None</td>
<td>21/12/2018</td>
<td>Active</td>
<td>10068185</td>
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<tr>
<td>10068192</td>
<td>MORSON PROJECTS LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>B200G7</td>
<td>Approval of Morson Projects Limited Change MP0014A for the introduction/relaunch of various antennas to support the later installation of a SATCOM System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval.</td>
<td>Installation in accordance with Morson Project Limited Modification Reference MP0014A. Instructions for Continued Airworthiness (ICA) in accordance with Morson Document ref MPL-C-02-AW006 Issue 1 dated November 2018. There is no Flight Manual/Change associated with this modification. O/S data is unchanged...</td>
<td>Prior to installation of this change it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>03/01/2019</td>
<td>Active</td>
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<tr>
<td>10068216</td>
<td>FOKKER SERVICES B.V.</td>
<td>REV. 1</td>
<td></td>
<td>757-300/ 757-200</td>
<td>ADS-B OUT on the Boeing 757 / This change introduces ADS-B capability to the Boeing 757 aircraft. The ATC transponders will be changed and they will interface with a suitable position sensor.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: Initial Issue of CS-ACN: CS-ACN 6.D.A.C.16; CS-ACN 6.D.A.C.25; CS-ACN 6.D.A.C.30; CS-ACN 6.D.A.C.35; CS-ACN 6.D.A.C.40...</td>
<td>The change is defined with Issue 1 of the Master Document A8767-ADSB-OUT-MDL-0001. The aircraft shall be operated in accordance with Revision 1 of the Flight Manual Supplement B757-ADSB-AMF-5/010 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>07/01/2019</td>
<td>Active</td>
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<td>10068217</td>
<td>FOKKER SERVICES B.V.</td>
<td>REV. 1</td>
<td></td>
<td>747-400/ 747-400F</td>
<td>ADS-B OUT on the Boeing 747 / This change introduces ADS-B capability to the Boeing 747 aircraft. The ATC transponders will be changed and they will interface with a suitable position sensor.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: Initial Issue of CS-ACN: CS-ACN 6.D.A.C.16; CS-ACN 6.D.A.C.25; CS-ACN 6.D.A.C.30; CS-ACN 6.D.A.C.35; CS-ACN 6.D.A.C.40...</td>
<td>The change is defined with Issue 1 of the Master Document A8747-ADSB-OUT-MDL-0001. The aircraft shall be operated in accordance with Revision 1 of Airplane Flight Manual Supplement B747-ADSB-AMF-5/010 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>15/01/2019</td>
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<td>10068267</td>
<td>AERODESIGN AND MANUFACTURING TECHNOLOGY S.L.</td>
<td></td>
<td></td>
<td>ATR 72-212 A</td>
<td>LOPA CONFIGURATION CHANGE FROM 70 TO 72 PAX ATR72</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval.</td>
<td>- Master Data List (M087-1721-AD03, Rev03); - Weight and Balance Manual Supplement (M087-1721- HBM5001, Rev03); - Illustrated Parts Catalog Supplement (M087-1721- IFCS01, Rev00) for the later installation of a SATCOM System.</td>
<td>Limited to MSN 982. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/01/2019</td>
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<td>STC Holder</td>
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<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>TUI AIRLINES BELGIUM N.V.</td>
<td>737-8</td>
<td>Wireless IFE system installation on B737MAX. The modification consists in the installation of a server and CWAIP. The system is operated by means of a control panel in the cabin.</td>
<td>The Certification Basis (CR) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Instruction for Continued Airworthiness: Doc no. JAF-DOA-ICAR-44M-0003REV00. Airplane Flight Manual Supplement: Doc no. AFMS-JAF-DOA-44M-0003 REV00. Master Data List: Doc no. JAF-STC-0062-0001 REV00. JAF-DOA-44M-0003REV00 or later revisions of the above listed document(s) approved/accepted/endorse...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>10/01/2019</td>
<td>Active</td>
<td>10068273</td>
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<td>ODA AVIATION GROUP, LLC</td>
<td>FAA STC ST08676AC</td>
<td>737-800</td>
<td>Installation of an equipment rack structure in the forward cargo compartment on Boeing 737-800 aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>ODA Aviation Group, LLC Master Drawing List No. R13859301, &quot;Equipment Rack Installation (Equipment Provisionals): Revision A, dated 1 May 2017,&quot; or later FAA approved revision. Required instructions for continued airworthiness are contained in Document No. R13859061, Revision IR, dated 24 October 2016...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>11/01/2019</td>
<td>Active</td>
<td>10068296</td>
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<tr>
<td>MARSHALL OF CAMBRIDGE AEROSPACE LTD</td>
<td>A318-112</td>
<td>Master Minimum Equipment List amendments for the Airbus A318 &quot;Elite&quot; Auxiliary Centre Tank System.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-MMEL Initial Issue. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>Certification Summary CS: A318/2042/17/0061 Rev.2, dated 08/10/2018; MMEL Supplement MMEL/A318 (1776)/10/0968 Rev.1, dated 30/06/2017; or later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>EASA STC 10035677 Rev.4 shall be embodied on the aircraft. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>11/01/2019</td>
<td>Active</td>
<td>10068301</td>
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<td>ETHIAD AIRWAYS ENGINEERING L.L.C.</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Low FrequencyULD Installation Referring to EASA mandate Commission Regulation (EU) 2015/2338 dated 1 Dec 2015, on underwater beacon installation prior to 1 Jan 2019. Thermal modification involves the installation of Dukane Seacom DK180 LowFrequency Underwater Locating Devices (60 days, 8.8 KHz, one unit per...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI-F-6318-004 Issue 1 Non-rechargeable LithiumBattery Installations - CS 25 at Amdt.21 for the affected area. The requirements for environmental protection and any other previously installed design change and/or repair it must be determined that the change/repair and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>- Master data list E30-1925031-0101 Issue 8 dated 13 Jan 2019; MMEL Supplement E30-1925031-PA11 Issue 3 dated 09 Jan 2019; MMEL Supplement E30-1925031-PA04 Issue 3 dated 09 Jan 2019; or later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>14/01/2019</td>
<td>Active</td>
<td>10068320</td>
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<tr>
<td>FOKKER SERVICES B.V.</td>
<td>REV. 1</td>
<td>747-400F</td>
<td>Non-activated GNSS Installation on 747-400F. Introduction of a Dual Non-Activated GNSS Unit and Antenna Installation on Boeing 747-400F.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment/CS-25, initial issue: 25.303</td>
<td>a</td>
<td>, 25.303</td>
<td>c</td>
<td>, 25.303</td>
<td>g</td>
<td>, 25.321</td>
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<td>STC Holder</td>
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<td>Foreign/NAI STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>REV. 1</td>
<td>T206H, T206H 208H</td>
<td>STC-AT-130, Casara 206 – Survey Configuration Modification of the cabin layout to a survey cabin configuration. Theware survey cabin configuration consists of three different surveymapping configurations (called as Configuration 1, 2, and 3) which eliminated above the hatch, in the baggage compart...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedrelatednoise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Approval Sheet (MD) Doc No.: STC-130-A-AS, Rev. 3</td>
<td>Heating Panels in accordance with the above list document(s) approved/accepted.</td>
<td>08/11/2020</td>
<td>Active</td>
<td>10068365</td>
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<tr>
<td>FACC OPERATIONS GmbH</td>
<td>A321-213/-214/-215/-216</td>
<td>A321-213/-214/-215/-216</td>
<td>A320-231/-232/-233</td>
<td>Passenger Luggage Space Upgrade for A320 Classic, OHSTC This change replaces the overhead stowage compartments doors with newdoors with a different shape to increase the overall volume of the overhead stowage compartment.</td>
<td>The Certification Basis for the original product as amended by thefollowing additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25 Amendment 20: CS25.305; CS25.306; CS25.307; CS25.314; CS25.326; CS25.356; CS25.603; CS25.605; CS25.606; CS25.61; CS25.63;</td>
<td>None</td>
<td>21/01/2019</td>
<td>Active</td>
<td>10068366</td>
<td></td>
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<td>AIRBUS S.A.S.</td>
<td>REV. 2</td>
<td>A321-21X/21Y/-21Z/21A</td>
<td>A321-21X/21Y/-21Z/21A</td>
<td>VASAT KA-BAND OAE INSTALLATION This modification covers the definition and installation of &quot;Outdoor Aerial Equipment (OAE)&quot; and associated wirings (from the OAE to theFAINS connected) Activation of the WiFi in cabin will be available later on under Minor Modification process. Revision 1: Adm...</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedrelatednoise and/or emissions levels of the product are unaltered and remain applicable to this certificate/approval w...</td>
<td>Certification Summary Report ref: A317-E11-0000 Issue 3, dated 10th Jan 2019- Modification Approval Sheet ref: MAS A0050 issue 1, dated 10th Jan 2019- Airplane Flight Manual Supplement: AFM, S MOD A0050 Rev.00, dated 11th Sep 2016- Maintenance Planning Document Supplement MPC: S MOD A0050Rev.0...</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce roadshow effect upon the airworthiness of the product.</td>
<td>10/11/2021</td>
<td>Active</td>
<td>10068413</td>
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<tr>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>REV. 1</td>
<td>FAA STO2036AL</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of ACSS NXT-600 Mode S Transponder for ADS-B Out with Optional ACSS NXT-600 GPS This AML STC installs the ACS/NXT-600 Mode-S Transponder with ADS-B Out functionality and optionally installs the ACSS NXT-600 GPS as a position source for ADS-B Out</td>
<td>The Certification Basis for the original product as amended by thefollowing additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS:ACS Subpart D. The requirements for environmental protection and the associatedrelated noise and/or emissions levels w...</td>
<td>None</td>
<td>23/01/2019</td>
<td>Active</td>
<td>10068416</td>
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<td>airplus engineering GmbH</td>
<td>REV. 1</td>
<td>PA-46R-350T (MATRIX)</td>
<td>PA-46R-350T</td>
<td>Installation of electrical Cabin Floor Heating Panels in accordance with Engineering Work Order EW-1059-00000.</td>
<td>The Certification Basis for the original product as amended by thefollowing additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS:ACS Subpart D. The requirements for environmental protection and the associatedrelated noise and/or emissions levels w...</td>
<td>Airplane Flight Manual Supplement PO-1995-00000, Issue 2.0, dated 21 January 2019</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>10/02/2021</td>
<td>Active</td>
<td>10068417</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NA STC ref.</td>
<td>Models</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10068437</td>
<td>SPAES GmbH &amp; Co. KG</td>
<td></td>
<td>PA-46-350P</td>
<td></td>
<td>Indium Antenna Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Master Document List Doc. No. MDL-064-147-53-01, Issue 0. Data 30.11.2018 for Continued Airworthiness Doc. No. ICA-064-147-53-01, Issue 0. Date 30.11.2018 for later revisions of the above listed document(s) approved/accepted/under the EASA System.</td>
<td>SJ/4 463/2011 only. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/01/2019</td>
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<td>10068480</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td></td>
<td>LEARJET MODEL 45</td>
<td></td>
<td>KA 94 WAAS GPS ANTENNA INSTALLATION</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>law. Northern Avionics MDL NAV0610, Rev. 0. To be maintained law. Northern Avionics ICA-NAV0610, Rev. 0. To be installed law. 8.NAV0610. Rev. 0. For later revisions of the above listed document(s) approved/accepted/under the EASA System.</td>
<td>None</td>
<td>25/01/2019</td>
<td>Active</td>
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<td>10068482</td>
<td>ETHIS AIRWAYS ENGINEERING L.L.C.</td>
<td>REV. 1</td>
<td>777-300ER</td>
<td></td>
<td>BOEING 777-300ER CABIN RECONFIGURATION REV 0 : Certifies Configuration 1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>- Master Document List, Ref. EB7-1525050-0L02, Issue 2. For later revisions of the above listed document(s) approved/accepted/under the EASA System.</td>
<td>Installation of this STC on an aircraft under EASA jurisdiction requires EASA Approved Cabin Configuration Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce ...</td>
<td>06/05/2019</td>
<td>Active</td>
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<tr>
<td>10068485</td>
<td>H4 AEROSPACE (UK) Ltd</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A321-211</td>
<td></td>
<td>44 B/C/D/E/+-, 106/100 Y/C/D/- Passenger Seat Reconfiguration</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>EASA Approved: H4 Aerospace Instructions For Continued Airworthiness No. H4A1896. 1EASA Accepted: H4 Aerospace Modification Summary No. H4A1896. Issue 2. For later revisions of the above listed document(s) approved/accepted/under the EASA System.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/01/2019</td>
<td>Active</td>
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<tr>
<td>10068513</td>
<td>365 AEROSPACE DDS LIMITED</td>
<td>REV. 2</td>
<td>ERJ 190-200 STD, ERJ 190-200 LR, ERJ 190-200 IGW, ERJ 190-100 STD, ERJ 190-100 LR, ERJ 190-100 SR, ERJ 190-100 IGW, ERJ 190-100 ECJ, ERJ 170-200 STD, ERJ 170-200 STD, ERJ 170-100 STD, ERJ 170-100 LR, ERJ 170-200 LR</td>
<td></td>
<td>Change to introduce New Cabin Interior Components Rev. 2: Transfer of STC from 365 Aerspace Ltd. to 365 Aerospace DDS Ltd.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>- 365 Aerospace Instructions for Continued Airworthiness – Change to introduce New Cabin Interior Components, Configuration 1 ICA-0529-051544, Issue A: 365 Aerospace Instructions for Continued Airworthiness – Change to introduce New Cabin Interior Components, Configuration 2 ICA-0529-051544, Issue A: 365...</td>
<td>Refer to 365 Aerospace Master Data List No. MDL-0629 for configuration standards that are a pre-requisite for this modification. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/08/2019</td>
<td>Active</td>
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<td>10068520</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td></td>
<td></td>
<td>Installation of Media Trolley for - ground charging configuration</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Description of Change ASK01-44 rev 3 Master Data List: see AML table in annex Operational Requirements document C-ACE815-0R rev 0. Swa Guide document C-ACE815-0U rev 0. To be installed law. 8.CAE815-0U-001 for later revisions of the above listed document(s) approved/accepted/under the EASA System.</td>
<td>Description of Change ASK01-44 rev 3 Master Data List: see AML table in annex Operational Requirements document C-ACE815-0R rev 0. Swa Guide document C-ACE815-0U rev 0. To be installed law. 8.CAE815-0U-001 for later revisions of the above listed document(s) approved/accepted/under the EASA System.</td>
<td>30/01/2019</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<tr>
<td>RUAG AG</td>
<td>Installation of Dual Garmin GNS-530 COMNAV/GPS System Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged.</td>
<td>Engineering Order TIA PC12-34.60.02EO, Revision A</td>
<td>This STC is approved only for the product configuration as defined in the approval design data referred to in the paragraph description. Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/02/2021</td>
<td>Active</td>
<td>10068529</td>
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<tr>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>A330-223F-243/243F A330-200-202-203-223 Installation of an airborne lightning and flight turbulence indication system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged.</td>
<td>Air France Change Approval Sheet CAS - A330-16-005 original issue dated 31 January 2018</td>
<td>Prior to installation of this change/repair the following interrelationship between this change/repair and any other previously installed or repaired product.</td>
<td>31/01/2019</td>
<td>Active</td>
<td>10068537</td>
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<tr>
<td>AERO-DIENST GMBH</td>
<td>Installation of a SDR Gateway Router including WiFi activation. The existing ICS-240 - ICS-240 Iridium Satellite Transceiver is replaced by the Iridium Direct Gateway Router (SDR). The SDR Gateway includes a dual channel Iridium transceiver and x30 / 40 module for high speed internet ground. A interface...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>MDL - 30300-DC-101 rev 00 - ICA - 20300-CLD-105 rev 00 - AFMS - 20300-CLD-106 rev 05</td>
<td>Limited to the following aircraft: 20124, 20125, 20128, 20134, 20139, 20143, 20146, 20148 and all subsequent serial numbers. Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/03/2019</td>
<td>Active</td>
<td>10068540</td>
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<tr>
<td>AERSALE, Inc.</td>
<td>Rattatoted Polyurethane Foam in B737NG Installation of an inherently electrostatically conductive Rattatoted Polyurethane Foam as an aircraft Ignition Mitigation Mean, in the centerline tank of Boeing 737-600, 737-700 and 737-800 model airplanes: Revision 1: Transfer of STC from Dassault Aircraft Service...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Master Data List B737NG- D0032-005, Revision IR, dated 03 March 2016</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/09/2019</td>
<td>Active</td>
<td>10068542</td>
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<tr>
<td>AEROTECH &amp; CONCEPT</td>
<td>Antenna relocation, ballast and cabin arrangement for subsequent installation of mission system...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>None</td>
<td>Configuration 1: The airplane must be previously equipped with In-Flight Entertainment System in accordance with Boeing Service Bulletin 757-34-0408 or equivalent</td>
<td>01/02/2019</td>
<td>Active</td>
<td>10068551</td>
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<tr>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>Installation of Honeywell RMA-55B Multi-Mode Receiver Rev. 1: Installation of Honeywell Multi-Mode Receivers (MMR). Configuration 1: Installation of 3 RMA-55B MMR part number (P/N) 066-50029-101 &amp; OPS PN HNRRBA-2601-001 and 2 GNSS antennas P/N 757-3438/0 071-50029-510 Configuration 2: Installation of...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Honeywell Master Drawing List 756-0265-001, revision B or later FAA approved revision 3, Honeywell Instructions for Continued Airworthiness (ICA) documentation 756-0265-800 revision 2 or later</td>
<td>Configuration 1: The airplane must be previously equipped with In-Flight Entertainment System in accordance with Boeing Service Bulletin 757-34-0408 or equivalent/Configuration 2: The airplane must be previously equipped with Installation of triple MMR (Honeywell RMA-55B MMR, 24-Channel, part number 066-50...</td>
<td>18/03/2020</td>
<td>Active</td>
<td>10068553</td>
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### List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<th>Description</th>
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<tr>
<td>CONCORDE BATTERY CORPORATION</td>
<td>SR02568LA</td>
<td>AS 332 L/AS 332 L2/EC 225 LP AS 332 CI AS 332 C1 AS 332 L</td>
<td>Replacement of existing SAGF main Battery models 40208-1, 40208-5, 40208-1 or 40208-5 with Concorde Battery Corporation Valve Regulated Lead Acid (VRLA) Battery models RG-332-1 or RG-332-2.Rev. 1 corrects typographical errors in the description of the STC.</td>
<td>Master Data List S-0012, Revision C, dated 1 March 2016, or later FAA approved revision of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of ELUASA Bilateral Agreement.</td>
<td>Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce a reducive affect upon the airworthiness of the product.</td>
<td>09/04/2021</td>
<td>Active</td>
<td>10086570</td>
</tr>
<tr>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>ST02665LA</td>
<td>757-200PF 757-200</td>
<td>Installation of Honeywell TRA-100B Mode S Transponders with ADS-B Out.</td>
<td>1. Honeywell Master Drawing List 756-0263-001, revision 2, dated 21 January 2018 or later FAA approved revision 2. Honeywell Airplane Flight Manual Supplement (AFMS) document number 756-0263-10, revision 2, dated 21 November 2017, or later FAA approved revision, OR Honeywell AFMS document number 756-0263-10, revision 2.</td>
<td>The installation of Honeywell EASA STC 10068553 must be incorporated on the aircraft prior to or at the time of this STC Installation.</td>
<td>05/02/2019</td>
<td>Active</td>
<td>10088895</td>
</tr>
<tr>
<td>BOGGI S.R.L.</td>
<td>ST03500LA</td>
<td>AS 350 B3 AS 350 B AS 350 B1 AS 350 B2 AS 350 B3 AS 350 BA</td>
<td>External Mirrors System (Mod N. 356P025)</td>
<td>356P0250-Mod-01 “Master Drawing List”, dated 04th February 2019; 356P0250-Mil-01 “Master Document List”, dated 04th February 2019; 356P0250-FMS-02 “FMS Mirror Installation on AS350/AS350C”, dated 04th February 2019 for later revisions of the above listed document(s) approved/accepted on behalf of the EASA is applicable.</td>
<td>Prior to installation of this change/repair it must be determined if the interrelationship between this change/repair and any other previously installed change and/or repair will introduce any adverse affect upon the airworthiness of the product.</td>
<td>08/02/2019</td>
<td>Active</td>
<td>10086065</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>SA09750DS</td>
<td>PA-46-500TP/PA-46-350T PA-46-500TP/PA-46-350P</td>
<td>Installation of an S-TEC 01326 Digital Flight Control System (DFCS).</td>
<td>1. Master Data List No. ST-967-AML-0001, Revision A, dated 18 September 2019(&gt;), or later FAA approved revision 2. Airplane Flight Manual Supplement, ST-967-AMS-0001, Revision - dated 20 September 2018 or later FAA approved revision 3. Instructions for Continued Airworthiness, ST-967-ICA-0001, Re-</td>
<td>1. Also eligible on models PA-46-310P and PA-46-350P when modified per STC EASA.0A-5.0218.2. Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce a reducive affect upon the airworthiness of the product.</td>
<td>05/02/2019</td>
<td>Active</td>
<td>1008611</td>
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<tr>
<td>PRO STAR AVIATION, LLC.</td>
<td>ST00310B</td>
<td>SEE FAA 05649-AML-101, REV. B.</td>
<td>Installation of a Honeywell CNS-X90 Cabin Network Accelerator and certification for portable electronic device (PED) tolerance.</td>
<td>1. Pro Star Aviation Installation Data List 05649-DL-101, Revision IR, dated 17 December 2018, or later FAA approved revision 2. Airplane Flight Manual Supplement (AFMS) listed in FAA Approved Model List 05649-AML-101, Revision B, dated 30 November 2018, or later FAA approved revisions. 3. Pro Star Avi...</td>
<td>1. Prior to installation of this design change it must be determined if the interrelationship between this design change and any other previously installed design change and/or repair will introduce a reducive affect upon the airworthiness of the product.</td>
<td>06/02/2019</td>
<td>Active</td>
<td>1008630</td>
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**Note:** This text is a representation of the information available on the EASA's website. The final version should be verified directly from the source.
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<th>STC Holder</th>
<th>Foreign/NAA</th>
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<td>BUGET INTERNATIONAL SALES AND SUPPLIES</td>
<td>FAA STC ST11057SC</td>
<td>737-700</td>
<td>Installation of an Executive Style Interior.</td>
<td>1. Certification Data List DL.185SM00, Revision F, dated 1 July 2013 or later FAA approved revision 2. Airplane Flight manual Supplement 1330021090, Revision 13, dated 24 June 2013 or later FAA approved revision 3. Instructions for Continued Airworthiness 1330050009, Revision C, dated 20 June 2013 or later.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2019</td>
<td>Active</td>
<td>1006842</td>
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<tr>
<td>AVIONIK STRAUBING ENTCWKUNGS GmbH</td>
<td>C90A, C90GT, C90GTI</td>
<td>Installation of Antenna Comant CI 401-490</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-23, Amendment 4-29, 251, 301, 303, 305, 307, 337, 385, 561, 571, 573, 575, 601, 603, 605, 607, 609, 611, 613, 625, 843...</td>
<td>- Instructions for Installation ASR-2018-019-K1-10, Revision 00. - Instructions for Continued Airworthiness ASR-2018-019-KCA-11, Revision 00.</td>
<td>Final compliance demonstration must be provided in the form of test installation for the equipment to be served by the system. The type inspection of the subject of this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previous change/repair and any applicable to this certificate/product.</td>
<td>08/02/2019</td>
<td>Active</td>
<td>1006860</td>
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<tr>
<td>FOKER TECHNIK B.V.</td>
<td>REV. 1</td>
<td>737-800</td>
<td>VIP Refurbishment RTAF Revision 1: Transfer of STC from Fokker Techniek B.V. (EASA.21J.059) to Fokker Techniek B.V. (EASA.21J.687)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/12/2019</td>
<td>Active</td>
<td>1006862</td>
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<tr>
<td>GULFSTREAM AEROSPACE Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>JetWave KA Band SATCOM System Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- GULFSTREAM AEROSPACE Ltd. Instructions for Continued Airworthiness No GAeL 18-29-ICA-501 Rev - (JetWave KA HSD System/Fin Cap Radome)-GULFSTREAM AEROSPACE Ltd. Instructions for Continued Airworthiness No GAeL 18-29-ICA-503 Rev - (Fin Cap Radome)-GULFSTREAM AEROSPACE Ltd. Aircraft Flight Manual.</td>
<td>None</td>
<td>13/02/2019</td>
<td>Active</td>
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<tr>
<td>32B SUPPORT SERVICES GmbH</td>
<td>EMB-505</td>
<td>Approval of the installation of Honeywell Aspire 200 SATCOM System and Honeywell &quot;GoDirect&quot;™ Router, including relocation of existing Indrinus Satcom Antenna.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>To be installed according to the Master Document LRMDA.131Z001 issue A dated 07th February 2019. To be operated according to EASA approved AMM supplement (EMB-505 Z0001 issue A dated 10th January 2019). To be maintained according to AMM Supplement AMM014G3120001 issue dated 15th Jan.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/02/2019</td>
<td>Active</td>
<td>1006868</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>1006892</td>
<td>BRITISH AIRWAYS PLC</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>Installation of Low Frequency Underwater Locator Beacon</td>
<td></td>
<td></td>
<td>08/02/2019</td>
<td>Active</td>
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<tr>
<td>1006872</td>
<td>GULFSTREAM AEROSPACE Ltd.</td>
<td>REV. 2</td>
<td>Latitude Technologies Satellite Data System Installation</td>
<td>The Certification Basis for the original product as amended by additional or alternative airworthiness requirements: Special Condition (s) CRI F-5.01 - Non-rechargeable Lithium Battery. Installations. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
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<td>06/04/2019</td>
<td>Active</td>
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<tr>
<td>10068718</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC ST04366AT</td>
<td>This STC is a replacement of the original equipment nickel cadmium main batteries manufactured by Raytheon with Concorde valve regulated lead acid main batteries.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST04366AT. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2019</td>
<td>Active</td>
<td>10068718</td>
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<tr>
<td>10068719</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC ST04273AT</td>
<td>Replace Original Main Battery Connector with Concorde Connector, Part Number 4000.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST04273AT. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2019</td>
<td>Active</td>
<td>10068719</td>
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<tr>
<td>10068724</td>
<td>HELI AUSTRIA GmbH</td>
<td>REV. 1</td>
<td>Installation of Garmin GTN 650 NAV/COM 1</td>
<td>The Type and OSDL Certification Bases for the original product as amended by the following: Additional or alternative airworthiness requirements: The following paragraph(s) are revised: ACN.B.VCS.601.C.S. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>This change is limited to VFR day and night operation. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product. The STC is valid for aircraft types and models as listed.</td>
<td>16/06/2022</td>
<td>Active</td>
<td>10068724</td>
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<tr>
<td>10068725</td>
<td>ADFIT AEROARCHITECTS</td>
<td>REV. 3</td>
<td>Ka SATCOM Antenna System Installation</td>
<td>The Type and OSDL Certification Bases for the original product as amended by the following: Additional or alternative airworthiness requirements - Equivalent Safety Finding(s): B-3301-001 Vibration and buffeting compliance criteria for large external antenna installation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/04/2024</td>
<td>Active</td>
<td>10068725</td>
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<td>STC Holder</td>
<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Associated Technical Documentation</td>
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<td>Description</td>
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<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>10068727</td>
<td>STC Holder</td>
<td>0</td>
<td>- Flight Manual Supplement AHUK350-FMS-122</td>
<td>&quot;Incompatibilities between EASA STC AHUK350-FMS-113 and FAA STC AHUK635-FMS-122&quot;</td>
<td>Installation of TRAKKA A800 Searchlight (Design Definition AHUK3503185-101)</td>
<td>12/02/2019</td>
<td>Active</td>
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<td>FAA STC</td>
<td>10068730</td>
<td>AVONICS SUPPORT GROUP, Inc.</td>
<td>0</td>
<td>1. Avionics Support Group</td>
<td>Certification Basis for the associated technical specification</td>
<td>Installation of an Avionics Support Group Constant FrictionMount(Mount) and</td>
<td>13/02/2019</td>
<td>Active</td>
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<tr>
<td>TCCA</td>
<td>10068744</td>
<td>BOOST HUMAN EXTERNAL CARGO SYSTEMS</td>
<td>0</td>
<td>FAA PMA P164277CE</td>
<td>FAA STC ST04247AT</td>
<td>FAA STC ST04247AT</td>
<td>29/08/2019</td>
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<td>PIEDMONT PROPULSION SYSTEMS, LLC</td>
<td>10068757</td>
<td>FAA STC</td>
<td>0</td>
<td>FAA PMA P0164277CE</td>
<td>FAA STC ST04247AT</td>
<td>FAA PMA P0164277CE</td>
<td>13/02/2019</td>
<td>Active</td>
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<td>RECAERO</td>
<td>10068768</td>
<td>CABIN RECONFIGURATION</td>
<td>0</td>
<td>FAA PMA P0164277CE</td>
<td>FAA STC ST04247AT</td>
<td>FAA PMA P0164277CE</td>
<td>13/02/2019</td>
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<td>STC No</td>
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<td>Rev.</td>
<td>FAA STC/STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
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<td>10068771</td>
<td>AERSALE, Inc.</td>
<td>REV 1</td>
<td>FAA STC ST0401NY</td>
<td>757-200F 757-200</td>
<td>ACS8 ADS-B OUT SYSTEM Installation of the ACS8 Automatic Dependent Surveillance - Broadcast (ADS-B) Out System in accordance with Aerospace Design &amp; Compliance, LLC. Master Data List.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aerospace Design &amp; Compliance, LLC. - AFM SUPPLEMENT ref. B753-AMF-0022 Rev. IR. Document ref. B753-AMF-0022 Rev. IR later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>As a prerequisite, the existing Transponder System must be in compliance with Mode S, Mode S ELTs, and Mode S EHS. Transponder Control Panel must provide an ATC/XPDR Failure Indication. (AC 20-168B refers). Aeronautics modified with this STC must be Operated in accordance with the AFM SUPPLEMENT ref...</td>
</tr>
<tr>
<td>10068773</td>
<td>KAMAN CORPORATION AIRCRAFT WHEEL AND BRAKE, LLC</td>
<td>FAA STC SAU4348CH</td>
<td>AVANTI AVANTI II</td>
<td>Installation of Parker Hannifin Corporation Aircraft Wheel &amp; Brake Conversion Kit, Part Number 159-283.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. There are requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>Document Number IM199-283, Kit Installation Publication; Main Wheel &amp; Brake Conversion Kit Parker Hannifin Part No. 159-283, Revision A, dated 10-October-2018, or later FAA approved revision and drawing number 50-166, Installation, Wheel &amp; Brake Revision A, dated 30 November 2017 or later FAA approved</td>
<td>Prior to installation of this design change it must be determined that the installation and/or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
<td>14/02/2019</td>
</tr>
<tr>
<td>10068808</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>REV. 3</td>
<td>FAA STC ST01685A</td>
<td>SEE FAA AML</td>
<td>Installation of Honeywell VHF 'Voice and/or Data Communications System NEW-PACE</td>
<td>The Certification Basis for the original product as amended: the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment for MDL configuration 1 CS25.1529 at amend. 0.</td>
<td>Honeywell Master Drawing List 159-0803-0001 Rev. R, dated 10 January 2019, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the installation and/or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
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<tr>
<td>10068835</td>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td>FAA STC ST01685AT</td>
<td>SEE FAA AML</td>
<td>Installation of Honeywell VHF 'Voice and/or Data Communications System NEW-PACE</td>
<td>The Certification Basis for the original product as amended: the following additional or alternative airworthiness requirements for the following paragraph(s) at a later amendment for MDL configuration 1 CS25.1529 at amend. 0.</td>
<td>None</td>
<td>None</td>
<td>None</td>
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<tr>
<td>10068836</td>
<td>HELI AUTRICIA GmbH</td>
<td>DC-9-82 (MD-82)</td>
<td>Installation of a fuselage ULB Installation of a fuselage Underwater Locator Beacon on MD-82</td>
<td>The Certification Basis for the original product as amended: the following additional or alternative airworthiness requirements Special Condition: CRS F-01 Non-rechargeable Lithium Battery Installations: The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>- Master Data List, MODE00420, MDL Initial issue Instruction for Continued Airworthiness - MODE00420, ICA1 Initial issue, later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>None</td>
<td>None</td>
<td>None</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA (IC) ref.</td>
<td>Movies</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10068840</td>
<td>VOYAGEUR AEROTECH, Inc.</td>
<td>TCGA-SA18-99</td>
<td>DHC-8-402, DHC-8-401, DHC-8-314, DHC-8-301, DHC-8-106, DHC-8-201, DHC-8-102, DHC-8-103</td>
<td>T360 Emergency Battery Power Supply Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition: CR F-191 LAN battery installations. The following paragraph(s) at a later amendment: CS 25 AMDT 17 CS25.1399(b), CS25.1351(c), CS25.1351(e), the...</td>
<td>- T360 Emergency Battery Power Supply Installation must be in accordance with True Blue Power Design Change Summary DCS 17-3225 Rev. NC. Operation of the aircraft must be in accordance with True Blue AFMS 17-3225 Rev NC. Maintenance modulie in accordance with True Blue Maintenance Manual Supplement...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/02/2019</td>
</tr>
<tr>
<td>10068894</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>A321-211/-231 A320-214/232-233</td>
<td>Direct View Camera System The design change is about the installation of a Direct View Camera System (DVCS), which is a system as designed by an additional system to improve the rearview direct view of cabin attendants in the AFT cabin on A320 family aircraft as replacement to the direct view cabin attendant seat...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification approval.</td>
<td>- Supplemental Type Definition Document: STDD25-18-09M issue 01 Rev. D dated 09 Dec 2019, Compliance Checklist: CCL25-19/122 Revision 1 dated 10 Dec 2019: Instructions for Continued Airworthiness: A-00-18-102-ICA Issue 1 Rev 3 dated 15 Oct 2019 for later revisions of the above listed documents...</td>
<td>This approval does not constitute an authorization for the installation of the modification &quot;A320 Direct View Camera System&quot; unless the individual cabin layout(s)&lt;,&gt; on which such modification is installed is demonstrated to be compliant to JAR25.785(h)(1) at Change 11. Furthermore, the embodiment of...</td>
<td>10/12/2019</td>
</tr>
<tr>
<td>10068940</td>
<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>FAA STC SA122DS</td>
<td>390</td>
<td>Installation of a Rockwell Collins Integrated Flight Information System (IFS-5000) into a Rockwell Collins Pro Line 21 system</td>
<td>The Certification Basis for the original product remains applicable to this certification approval, except where amended by additional or later amendments to FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and are applicable to this certification approval.</td>
<td>* Master Data List: Premier 1 (IFS) Premier II (IFS) and XM&lt;&gt;&lt;&gt; &lt;,&gt;Document: 620-00-0011 Revision IR dated 28 November 2018 or later FAA approved revision. Instructions for Continued Airworthiness: Document 620-00-0026 Revision A dated 19 November 2018 or later FAA approved revision. Airplane Flight Manual Supplement...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and or repair will introduce no adverse effect upon the airworthiness of the product. Must have factory installed Pro Line 211 and EASA...</td>
<td>26/02/2019</td>
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<tr>
<td>10068941</td>
<td>JET AVIATION AG</td>
<td>17-3225 Revision 1.</td>
<td>MYSÈRE-FALCON 20 - FS MYSTERE-FALCON 20 - ES MYSTERE-FALCON 20 - DS MYSTERE-FALCON 20 - CS</td>
<td>FMS Replacement Replacement of a single Honeywell INS-XLS Typ single Universal INS-ULIncorporating a SBAS GPS receiver Removed: Honeywell INS-XLS GPS Antenna Installed: Universal Avionics INS-1Lx FMS Universal Avionics SBAS GPS Antenna Universal Avionics Solid State Transfer Unit (SSSTU)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: JBSL has elected to comply with later requirements of EASA CS25 amendment 21 for paragraphs as follows: 301, 303, 305, 307,...</td>
<td>- Master Data List: JBSL-M-0118MDL-01, Issue 1&lt;&gt;&lt;&gt; &lt;,&gt;Airplane Flight Manual Supplement: JBSL-M-0118AFM-01, Issue 2&lt;&gt;&lt;&gt; &lt;,&gt;Instructions for continued airworthiness: JBSL-M-0118GCA-I-01, Issue 2&lt;&gt;&lt;&gt; &lt;,&gt;later revisions of the above listed documents...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product. Must have factory installed A-00-18-102-ICA Issue 1...</td>
<td>26/02/2019</td>
</tr>
<tr>
<td>10068943</td>
<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>REV. 3</td>
<td>FAA STC SA1219SC</td>
<td>S25B 525, 525A 525B</td>
<td>Rockwell Collins Pro Line Fusion Installation Rev 1 introduces software version that resolves the ACAS Fly-To Cueing issue addressed by EASA AD 2019-0179 Rev. 3 adds S25 and S25A model into the scope of this change and adds COPDLC functionally for S25 and S25A model.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): For S25, S25A (525B): CR F-69 Weather Displays For S25, S25A: SC F23-1437-01 Flight Recorders (including Data Links Recording)...</td>
<td>The following...</td>
<td>Mobile Enablement System is automatically inhibited in flight for EASA/registered aeroplanes Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect...</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC No.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>HONEYWELL INTERNATIONAL INC.</td>
<td>FAA STC ST01592AT</td>
<td>A300 F4-622 R, A300 F4-605 R, A300 CA-605 R VARIANT F, A300 B4-622 R, A300 B4-622, A300 B4-620, A300 B4-605 R, A300 B4-603, A300 B4-601</td>
<td>Installation of an AlliedSignal Mark II ARACS</td>
<td>The Certification Basis for the original product remains applicable to this certificate/revision, except where amended by additional or later alterations if indicated on FAA STC. There are requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>Master Drawing List No. 159-08175-0001, Revision D, dated 7 April 2000 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/02/2019</td>
<td>Active</td>
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<tr>
<td>AEROSERVIS, S.R.O.</td>
<td></td>
<td>L 410 UVP-E20, L 410 UVP-E</td>
<td>Installation of the Garmin GTN 725 with enabled TAWS A option, and an navigational information source to comply with PBN (RNAV) and LPV(Approach)</td>
<td>CS-ACS Initial Issue. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/revision.</td>
<td>Installation in accordance with Aeroservis Aircraft Maintenance Modification ref AE117 and Master Document List ref AE 177006X_RA, dated, 2. 10. 2018. Operate in accordance with Aeroservis Aircraft Flight Manual Suplement/ref LET 410 UVP-E, GTN 725 with TAWS-A, or L 410 UVP- E20, GTN 725 with TAWS-A...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/02/2019</td>
<td>Active</td>
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<tr>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>MYSTÈRE FALCON 900 (MF900) FALCON 900EX (F900EX)</td>
<td>Modification DFS 9872 Issue A: Portable Emergency Locator Transmitter (ELT) Installation.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative environmental requirements: Special Condition (s): F-6128-001: Non-rechargeable LiThium Battery/Sensors. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/revision.</td>
<td>MASTER DOCUMENT LIST Document Ref. DFS-DE-17-F9-9872 Issue A dated 2 February 2018 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/02/2019</td>
<td>Active</td>
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<tr>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 1</td>
<td>747-400</td>
<td>Automatic Dependent Surveillance – Broadcast (ADS-B) OUT (DO-260B) activation in 747-400 aircraft. Revision 1 of this STC expands the approved design change configuration by allowing installation of additional Multi-mode Receiver and Air Traffic Control (ATC) Transponder/Part Numbers as defined in A...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/revision.</td>
<td>Air France Service Bulletin AFASB-B747-18-001-34-001 original issuance France Service Bulletin AFASB-B747-16-001-34-002 original issuance France Airframe Flight Manual Supplement AFASB-B747-18-001-001 Revision 14 Air France Change Approval Sheet/CAS-B747-16-001 Revision 2 or later revisions of the above Iss.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/11/2019</td>
<td>Active</td>
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<tr>
<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>REV. 2</td>
<td>FAA STC SA11222DS</td>
<td>SEE FAA AML</td>
<td>Installation of automatic dependent surveillance-broadcast (ADS-B) Out (DO-260B) Collins Pro LineFusion. Textron 525, 525A and 525B with the applicable avionics/model combinations for GPS-4000S-101 and updates applicable/environments/model/ combinations for Textron 525, 525A and 525B with the Collins Pro LineFusion.</td>
<td>Master Data List 583-03-0001. Revision IR, dated 23 January 2018, or later FAA approved revision: For Instructions for Continued Airworthiness and Aircraft Flight Manual Suplement reference the AML.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and models approved...</td>
<td>18/01/2021</td>
<td>Active</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10069007</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>REV. 1</td>
<td>BO 105 S</td>
<td></td>
<td>Parcel Load Arm for Side Step of BO105: Modification of a load arm on an BO105 left hand side step including forward and rearward camera position. Rev. 1: Transfer from Parrel Aerospace Systems Europe GmbH to AirborneTechnologies GmbH.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unaltered and remain applicable to this certificate/ approval.</td>
<td>CER_T0700_0000217 Certification ProgramDOC_17-40_0308 2017 Declaration of ComplianceCA-17-040_03 082017 Instructions for continued AirworthinessAFM_17-040_08112017 Flight Manual Supplement: later revisions of the above listed documentation approved/ accepted Real time the EASA system</td>
<td></td>
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<tr>
<td>10069009</td>
<td>DART AEROSPACE Ltd.</td>
<td></td>
<td>TCCA STC SH18-50 ISSUE 1</td>
<td>EC 130 B4/ EC 130 T2</td>
<td>Emergency Float System Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>Master Document List MD130-4 Revision A, dated 15 November 2018Installation Instructions ICA130-4 Revision B, dated 21 October 2018Instructions for Continued Airworthiness ICA130-4;Revision 0, dated 12 June 2019Flight Manual Supplement RFM130-2 Revision A, dated 11 October 2018for later revisions of L.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10069010</td>
<td>LIFEPORT, Inc.</td>
<td></td>
<td>FAA STC SA2234LA</td>
<td>B300C</td>
<td>Installation of a ServiPlex.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC. There are no requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>1. Master Drawing List for the ServiPlex installation in the HawkerBeechcraft B300C, MLD-849-00, Revision, dated 29 March 2009, or later FAA approved revision.2. Airplane Flight Manual Supplement No. EGR-848-11, dated 13 May 2009 or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10069011</td>
<td>Q.C.M. DESIGN GmbH</td>
<td></td>
<td></td>
<td>BD-700-1A10</td>
<td>Change of aircraft LOPA and installation of SATCOM system</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): F-O1 Security Protection of Aircraft Systems and Networks; Deviation(s): D-01; Doors between Passenger Compartments Installation of a go-pro 2Gc Cabin Connectivity SystemRev. 1: corrects a typo: Rev. 2: amends the installation with demand option of Integral Cabin Connectivity System.</td>
<td>The change is defined with the initial issue of Q.C.M. Certification Report BD77- QC055-015;The aircraft shall be operated in accordance with Aircraft Flight ManualSupplement BD77-QC055-015-M1- AFM. This AFM consists of two parts. 1. BD77- QC055-015-M1-AFM titled LOPA Change. Revision 022. BD77-QC055-015...</td>
<td>The Luftfahrt Bundesamt Joint Aviation Authorities Supplemental TypeCertificate LBA.21/n2.7A00575, Installation of complete custom aircraft interior shall be incorporated prior to or in conjunction with the STC Prior to installation of this change/repair it must be determined that the interrelationship...</td>
</tr>
<tr>
<td>10069012</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST03417NY</td>
<td>A321-272NX A321-253NX-27IN-271NX A321-253NX-25IN-252NX X-253NX A321-213/231/232/251NX A321-111/112-131/1-211- 212 A320-232/232/251NX-252 A320-213/212/214/216- 231 A319-131/-120/133/ A319-131/-120/133 A318-111/112/114/- 115</td>
<td>Installation of a go-pro 2Gc Cabin Connectivity SystemRev. 1: corrects a typo: Rev. 2: amends the installation with demand option of Integral Cabin Connectivity System.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CS XX paragraphs + amdt The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Delta Engineering Master Data List 0164-0076-2406 RevisionBP, dated 22 September 2022(+) (+) + instructions for ContinuedAirworthiness, Document No. 0784-09097-2406; Rev. O, dated 22 September 2022(+) (+) + Weight and Balance Data: D12-716-30015 Rev B SWIV(+) (+) or later revisions of the above listed document.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. Operational approval to use to use T-PEDs musts...</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/ NAA STC ref.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>10069040</td>
<td>STC Holder System</td>
<td>2180</td>
<td>2180</td>
<td>Increase of limited mass for sailplane. With this change the maximum allowed sailplane mass which can be towed increased from 580kg to 800kg.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Special Condition(s): CRI-0-09 Gider towing. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AFM supplement Amend Varga 2180 rev 0 dated 23 March 2018</td>
<td>Maintenance Manual supplement Doc. No. E-17GTSMM rev 0 dated 23 March 2018</td>
<td>Installation Instructions Doc. No. E-17GTSIS rev 0 dated 23 March 2018 Later revisions of the above listed documents approved by EASA.</td>
</tr>
<tr>
<td>10069041</td>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td>604</td>
<td>CL-600-2B16 (604) VARIANT, CL-600-2B16 (601-3R) VARIANT) , CL-600-2B16 (601-3A) VARIANT, CL-600-2A12 (601 VARIANT) , CL-600-14A1 (600)</td>
<td>Lithium Ion Powered Devices &amp; Equipment to approve the carriage of Non Rechargeable Lithium Ion battery power/electronics/emergency equipment on board the aircraft type.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List - MOOE0257_MDL Rev. 1</td>
<td>Initial Issue/In-Instruction for Continued Airworthiness - MOOE0257, ICA9 Rev. 1</td>
<td>Ikhana Group, LLC.</td>
</tr>
<tr>
<td>10069042</td>
<td>IKHANA GROUP, LLC</td>
<td>REV. 1</td>
<td>FAA STC SA0245MY</td>
<td>DHC-6 SERIES 400</td>
<td>Installation of R. W. Martin Air Conditioning System Rev. 1</td>
<td><em>R. W. Martin Inc. Drawing List, Report ER28-2602V, Revision C, dated 7th September 2010, FAA approved 9 September 2010 or later FAA approval revision.</em></td>
<td>For instructions for Continued Airworthiness refer to <em>R. W. Martin</em>Instructions for Continued Airworthiness. Document No. AC-64-11 Revision (R), dated 40.</td>
<td>Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10069048</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, LLC. (ACS)</td>
<td>REV. 1</td>
<td>FAA STC ST036GTLA</td>
<td>777-200-200LR/-300/-300</td>
<td>Modification to the existing Mode S Transponder system to add NEXTGENADS-B Out and optionally install the ACSS NAO-900 GPS as an ADS-B Outposition source.</td>
<td>FAA approved Index of Master Data Lists, Document No. 6900077, Revision 2, dated 17 August 2016, or later FAA approval revision.</td>
<td>The installation of the optional ACSS NAO-900 GPS as an ADS-B Outposition source, is not permitted. Prior to installation of this design change it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>07/03/2019</td>
</tr>
<tr>
<td>10069103</td>
<td>CESSNA DUESSELDOFF CITATIONSERVICE CENTER GmbH</td>
<td>525A</td>
<td>Upgrade GNS-430/530(A) to GNS430/530A(W)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List, Report 2018-21-16, Revision B, dated 5 March 2019. Installed according to Installation Procedure, Report 2018-21-1, Revision A, dated 15 August 2018. Operated according to Airplane Flight Manual Supplement, Report 2018-21-4, Revision IR, dated 21 August 2018. Maintained according to CRI O-09.</td>
<td>Limited to the following S/N: 525-0001 through 525-0002, 525A-0001 through 525A-009. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2019</td>
<td>Active</td>
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<tr>
<td>10069118</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 3</td>
<td>A321-211/-231 A320-214/-230/-233</td>
<td>Alternative aluminium material at frame G70/Alternative aluminium material for the drag struts and angles at frame F370/Isolated. Additionally, the material thickness has been increased from 2.5mm to 2.6mm and all bend radii have been increased from 4mm to 5mm ±1mm. Revision 2: extension of the applicable...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. Additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment, as per CL601012-A0011 Rev. 0, the requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>L090120-MDL-0010 rev. 0&lt;sup&gt;1&lt;/sup&gt;</td>
<td>Limited to S/N VAC 158-80 and VAC 180-82 Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>04/11/2019</td>
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*Note: Some text has been replaced with placeholders due to the limitations of the text extraction process.*
<table>
<thead>
<tr>
<th>STC Holder</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>E A D AEROSPACE</td>
<td>A320-233 Installation of ADS-B Out System with active limitations/conditions</td>
<td>Master Data List EAD-10068894</td>
<td>For the installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/03/2019</td>
<td>Active</td>
<td>10068894</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>Installation of Pratt &amp; Whitney Engine Services, Inc. Aircraft Data Acquisition System ADAS</td>
<td>Installation Manual DAAS-G-050-4, Revision B. dated 26 April 2007, or later FAA accepted revision.</td>
<td>For the installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/03/2019</td>
<td>Active</td>
<td>10068514</td>
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<td>A320-233</td>
<td>Installation of Pratt &amp; Whitney Engine Services, Inc. Aircraft Data Acquisition System ADAS</td>
<td>Installation Manual DAAS-G-050-4, Revision B. dated 26 April 2007, or later FAA accepted revision.</td>
<td>For the installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/03/2019</td>
<td>Active</td>
<td>10068514</td>
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<tr>
<td>ELLIOTT AVIATION</td>
<td>FAA STC 5600L Installation of ADS-B Out System with Shad in Avionics AMC</td>
<td>- Compliance Checklist: CCL25-15/STM Issue 0 dated 21 October 2019, or later FAA approved revision.</td>
<td>For the installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/10/2019</td>
<td>Active</td>
<td>10068514</td>
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<tr>
<td>LUTHANSA TECHNICA AG</td>
<td>A320-233 Installation of Pratt &amp; Whitney Engine Services, Inc. Aircraft Data Acquisition System ADAS</td>
<td>Installation Manual DAAS-G-050-4, Revision B. dated 26 April 2007, or later FAA accepted revision.</td>
<td>For the installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/03/2019</td>
<td>Active</td>
<td>10068514</td>
</tr>
<tr>
<td>PRATT &amp; WHITNEY ENGINE SERVICES, IN</td>
<td>FAA STC 5600L Installation of Pratt &amp; Whitney Engine Services, Inc. Aircraft Data Acquisition System ADAS</td>
<td>Installation Manual DAAS-G-050-4, Revision B. dated 26 April 2007, or later FAA accepted revision.</td>
<td>For the installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/03/2019</td>
<td>Active</td>
<td>10068514</td>
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<tr>
<td>STC Holder</td>
<td>System</td>
<td>Limitations/Conditions</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>PRATT &amp; WHITNEY ENGINE SERVICES, IN</td>
<td>FAA STC ST002580 560XL</td>
<td>Installation of Pratt &amp; Whitney Engine Services, Inc. Data Transmission Unit (DTU)</td>
<td>The Certification Basis for the original product remains applicable. Modifications are approved, except where mandated by additional requirements or emissions levels of the original product.</td>
<td>Installation Manual DTU-2014.5-105-4, Revision B, dated 26 April 2007, or later FAA approved revisions.</td>
<td>13/02/2019</td>
<td>Active</td>
<td>10069157</td>
</tr>
<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>B200GT</td>
<td>EGPSWS inhibition of alert during mission operation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with AEROTEC Concept Modification ref 18099s defined in Master Document List (MDL) ref, AE18099ML Rev 0, dated 04.03.2019. Operation in accordance with AEROTEC Flight Manual Supplement (AFM) ref, AE18099AFM, Initial Issue, dated March 2019.</td>
<td>12/03/2019</td>
<td>Active</td>
<td>10069165</td>
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<tr>
<td>SAFRAN SEATS</td>
<td>747-400</td>
<td>LGW Aircraft Zone A and Zone B reconfiguration</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Heath Tecna Definiton Document R222 Rev. 1, dated 22 October 2004, or later FAA approved revisions. Instructions for Continued Airworthiness HTM911, Initial Release, dated 19 August 2004, or later approved revision.</td>
<td>13/02/2019</td>
<td>Active</td>
<td>10069180</td>
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<tr>
<td>SAFRAN SEATS</td>
<td>737-600</td>
<td>Introduction of a new 108 Y class configuration along with galleys and wardrobe as defined in Heath Tecna Drawing B-04196 and associated DDPR660</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Heath Tecna Modification Document C344 and associated DOP, dated 24 March 2005, or later approved revision.</td>
<td>12/03/2019</td>
<td>Active</td>
<td>10069191</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST0435NY 777-200LR 777-300</td>
<td>Installation of a Bruce Aerospace Lighting</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where mandated by additional requirements or emissions levels of the original product.</td>
<td>Delta Engineering Master Data List 0743-10360-2423 Revision F, dated 18 May 2018 or later FAA approved revisions to Delta Engineering Master Data List previously listed.</td>
<td>12/03/2019</td>
<td>Active</td>
<td>10069191</td>
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<tr>
<td>STANDARD AERO</td>
<td>FAA STC ST04263CH GV-SP GV GV-IX GII GII (G-1158A) GII (G-1159) GIII (G-1198B)</td>
<td>Activation of a WiFi System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where mandated by additional requirements or emissions levels of the original product.</td>
<td>1. Standard Aero Master Data List 103389, Revision A, dated 3 May 2018 or later FAA approved revision. 2. FAA Approved Airplane Flight Manual Supplement document number 103456Rev 2A (V) dated 3 May 2018, or later FAA approved revision.</td>
<td>13/02/2019</td>
<td>Active</td>
<td>10069211</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>10069212</td>
<td>STANDARD AERO</td>
<td>ST03274CH</td>
<td>BD-700-1A11</td>
<td>Activation of an 802.11 a/b/g/n/ac WiFi System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST03274CH. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Standard Aero Master Data List 1005054(&lt;&gt;)</td>
<td>Prior to installation of this design change it must be determined if there is any adverse interrelationship between this design change and any other previously installed design change and/or repair will introduce new adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10069214</td>
<td>2 EXCEL DESIGN LIMITED</td>
<td>CA3-C-EXCEL</td>
<td>277-200F</td>
<td>Installation of Low Frequency Underwater Locating Device onto Boeing 727</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (s): CRI F-01 Non-rechargeable Lithium Battery/Installations. The following paragraph(s) at the latest amendment: CS 25.17.(e)(1)(&lt;&lt;(),&lt;&gt;)</td>
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<tr>
<td>10069215</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>300, 300C</td>
<td>Installation of External Electro-Optical-Infrared (E/OIR) sensor, Search and Rescue antenna (SAR-UFF) and Radome</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, AE17114MDL, Rev. 0; Certification program/ presentation dossier, AE17114PCDP, Rev. 1; Aircraft Flight Manual Supplement, AE17114AMS, Rev. 0; Instructions for continued airworthiness, AE17114ACA, Rev. 0; or later revisions of the above listed document(s) approved/applicable under the EASA system.</td>
<td></td>
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</tr>
<tr>
<td>10069222</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>ST03434NY</td>
<td>777-200</td>
<td>Installation of a Cabin Interior Reconfiguration.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST03434NY. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Delta Engineering Master Data List 0143-10259-2486 Revision D dated 24 April 2018 or later FAA approved revisions to Delta Engineering/Master Data List previously listed.</td>
<td>Prior to installation of this design change it must be determined if there is any adverse interrelationship between this design change and any other previously installed design change and/or repair will introduce new adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10069231</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A340-541, A340-642</td>
<td>Installation of C-Music System. The change includes the Post Structure Assembly, (PSA) housing the main components of the system. The PSA (cable) is mounted at the lower section off the fuselage.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Findings(s): 8-101 Vibration buffeting compliance criteria for large external antenna installations. The requirements for environmental protection and the a...</td>
<td>The change is defined with Revision B of Master Data List Document 6418-340-541-1-5300T The aircraft shall be maintained in accordance with Revision B of Aircraft Flight Manual Supplement AFMS-340-541-1-5300T. The aircraft shall be maintained in accordance with Revision B of Aircraft Flight Manual Supplement AFMS-340-541-1-5300T.</td>
<td>Prior to installation of this change/repair it must be determined if there is any adverse interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>10069246</td>
<td>ADAC LUFTFAHRT TECHNIK GmbH</td>
<td></td>
<td></td>
<td>EC135 P3, EC135 T3</td>
<td>Night Vision Imaging Systems (NVIS) Certification of Mission Equipment</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 27 Amendment 4 for the following paragraphs: 27.1: 27.140(i), 27.601, 27.771(a), (b): 27.173, 27.1306(a), (b), 27.1209 (a)...</td>
<td>- Airbus A330 Flight Manual Supplement EASA STC no. EC135-0368 256-MFS Revision 0 - Master Document List EC135-0368-001-LMD Revision 0 - Installation Instructions EASA STC EASA STC 10069246 Revision 1 - Maintenance Manual Supplement EC135-0368-023-MMS Revision 0 - later revisions of the above listed document(s) approved...</td>
</tr>
<tr>
<td>10069248</td>
<td>BROMMA AIR MAINTENANCE AB</td>
<td></td>
<td></td>
<td>DORNIER 228-212</td>
<td>Installation of a belly pod</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Bromma Air Maintenance AB modification 17-27 and Bromma Master Data List ref MDL-17-27, Rev A dated 08.03.2019 Operation in accordance with Bromma Air Maintenance AB Flight Manual Supplement ref FMS-17-27, Rev A, dated 04.12.2018 Maintain in accordance with Bromma AIC...</td>
</tr>
<tr>
<td>10069253</td>
<td>ANORA INTERNATIONAL LLC</td>
<td></td>
<td>FAA STC ST01563A</td>
<td>747-200B/-200C/-200F/30-400</td>
<td>Installation of Lower Deck Ball Mats</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product...</td>
<td>FAA approved Ancra International LLC Controlled Drawing List DL82900 dated 10 October 2003, or later FAA approved revision.</td>
</tr>
<tr>
<td>10069255</td>
<td>BEONE HAMBURG GmbH</td>
<td></td>
<td>C90A, C90GT, C90GTI</td>
<td>C90A, C90GT, C90GTI</td>
<td>Installation of L-Band Antenna Installation of L-Band Communication Antennas</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/approval.</td>
<td>- Installation of L-Band Antenna in accordance with BEone Master Document List. 617-1005-MDL-001, at revision A, dated March 2019; Instructions for Continued Airworthiness, BEone document 617-1005-ICA-001, at revision A, dated March 2019 or later revisions of the above listed document(s) approved...</td>
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<tr>
<td>10069256</td>
<td>AIRBUS DEFENCE AND SPACE S.A</td>
<td></td>
<td></td>
<td>A330-243</td>
<td>Conversion of A330-200 Series Aircraft to A330 Multi Role TransportTanker MMTF (Multi National MRFF Fleet)- Rev. 1 introduces NEDEVAC Configuration via change DCV0934-E; Rev. 2 updates the change reference- Rev. 3 updates the AFM reference- Rev. 4 introduces change DCV0934-E, AFM Limitation for the...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (a) D-4318-002, Air Medical Service - The following paragraph(s) at a later amendment: CS 25.0562 Initial issue for affected areas by this STC/CS 25.085...</td>
<td>- CONVERSION DEFINITION A330 MMTF STC document ref. TAE-FA-DM-170856/Issue A - Flight Manual Supplement EASA APPROVED MODEL A-330 MMTF AIRPLANE FLIGHT MANUAL documentRef. TEA-FA-DM-170852, including Weightand Balance limitations reference TAS-FM-TD-19-001 Limitations off Weight and Balance...</td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>10069257</td>
<td>AVONIK STRAUING ENTWICKLUNGS GmbH</td>
<td>1</td>
<td>10069257</td>
<td>PA-31T (CHEYENNE/ CHEYENNE II) PA-31T3 (CHEYENNE III) PA-31T1 (CHEYENNE A) PA-31P (PRESSURIZED NAVAJO) PA-31P-350 (MOJAVE)</td>
<td>Garmin G3X8114 and/or Comert C1005-6 installation of Garmin G3X8114 autothrottle and/or Garmin C1005-6 XDRP diversity antenna on top fuseage.</td>
<td>FAA STC</td>
<td>Installation defined in Master Document List, reference Doc. N &quot;ASR-2018-070-MDL-01-01, dated 11/03/19revisions for installation, reference Doc. N 'ASR-2018-070-MDL-01-00 dated 11/03/19&quot;</td>
</tr>
<tr>
<td>10069280</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>1</td>
<td>10069280</td>
<td>FAA STC SAF007LA</td>
<td>SEE FAA AML Modification to upgrade the existing Mode S Transponder system by installing the ACSS NKT-700 Transponder with ADS-B Outdated as optional install the ACSS NKG-500 GPS as the position source for ADS-B Out.</td>
<td>FAA approved AML STC Master Data List (MDL) as defined in the AML No. SA0007LA dated 15 January 2019, or later FAA approved revision FAA approved Airplane Flight Manual Supplement and Instructions for Continued Airworthiness are defined in the AML No. SA0007LA dated 15 January 2019, or later FAA approved revision.</td>
<td>Eligible airplane models must be listed on the FAA Approved Model List (AML) for this STC to be applied. Prior to installation of this design change it must be determined that there will be no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10069291</td>
<td>BELL TEXTRON, Inc.</td>
<td>REV. 1</td>
<td>10069291</td>
<td>FAA STC SR0463NY</td>
<td>407 Installation of a GPMIX Foreight MX Condition Monitoring System (CMS) Revision 1 adds Bell 407 Analog Aircraft.</td>
<td>FAA approved Model List (AML) No. SR0463NY-AML as amended on 6 December 2022; Instructions for Continued Airworthiness (ICA) as specified in the SR0463NY-AML- AFM. Rotorcraft Flight Manual Supplement (RFMS) as specified in the SR0463NY-AML or later revision.</td>
<td>Prior to installation of this design change it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10069305</td>
<td>L2 CONSULTING SERVICES Inc.</td>
<td>1</td>
<td>10069305</td>
<td>FAA STC ST10095SC</td>
<td>757-200, 757-200P, 757-300 Installation of an Iridium Communication System (ICS)</td>
<td>Master Drawing List 2325-B47-MDL, Revision L dated 8 December 2018, or later FAA approved revision Airplane Flight Manual Supplement 2325-B47-AM. Revision D dated 18 December 2018.</td>
<td>Prior to installation of this design change it must be determined that there will be no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Introduction of Structural Provisions For A JET AVIATION AG Limitations/Conditions</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>LUFTHANSA TECHNOLOGIES</td>
<td>This design change consists of the installation of an Auxiliary Equipment Rack and Provisions to integrate avionic and electrical equipment into a fixed lower forward cabin compartment area.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>20/03/2019</td>
<td>Active</td>
<td></td>
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<tr>
<td>STC TWENTY ONE LIMITED</td>
<td>Installation of two Securaplane 24 VDC 29 Ah sealed Lead Acid Batteries</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>21/03/2019</td>
<td>Active</td>
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</tr>
<tr>
<td>STC TWENTY ONE LIMITED</td>
<td>Introduction of Structural Provisions For A No.3 VHF Communications Antenna</td>
<td>This design change (S21.53-13-1056) is an Alternative Method of Compliance (AMOC) to Airworthiness Directive AD 2013-04-05 in the regionively applicable to this certificate/approval.</td>
<td>None</td>
<td>21/03/2019</td>
<td>Active</td>
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<tr>
<td>JET AVIATION AG</td>
<td>This STC introduces several changes to the existing GVI (G650) cabin layout, which has been approved under FAA STC ST04252AT-D and EASA STC 10045299. The modifications include, amongst others, the replacement of the LHF three seats dais by two single seats, the relocation/installation of folding tables...</td>
<td>The Certification Basis for the original product as amended by this STC is limited to MSN 6115 only.</td>
<td>- Master Data List JESL - M-015004DH-01, Issue 01- ALS in Chapter 4 of JASU-L- M-01504DH-01, Issue 01 or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>22/03/2019</td>
<td>Active</td>
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<tr>
<td>STC no.</td>
<td>STC Holder</td>
<td>Module</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10069380</td>
<td>CRANFIELD AEROSPACE SOLUTIONS LIMIT</td>
<td>A109E</td>
<td>Installation of LineFusion® FTI</td>
<td>System installation of flight test instrumentation system to provide the ability to monitor aircraft systems and gather independent data without control or feedback from the FTI system into any aircraft system. The system includes monitoring, processing, storage and transmitting data.</td>
<td>The Certification Basis (CB) for the original product as amended applies to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/approval.</td>
<td>- Suplementary STC Definition Document STDC03-155MA Issue 04 rev 00</td>
<td>21/06/2024</td>
</tr>
<tr>
<td>10069399</td>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>A350-223</td>
<td>IFE upgrade in Y-Claw Air Greenland has awarded IRF Technics to provide an IFE upgrade/installation of new IFE equipment on existing business class (30) economy class (248) seats. The business class seats were upgraded with IFE with STC 10064829. This STC application is to upgrade the existing system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/approval.</td>
<td>- Prior to the installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/06/2019</td>
</tr>
<tr>
<td>10069404</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>DR-400 RP, DR-400/100</td>
<td>Removal of VHF Antenna System, as installed per STC 10064496 (up toiffer 3) and 10040512 (initial issue) - Option 1 and Option 2Rev: 1: Updated ICA (including updated inspection intervals). AMOC to ADUS-2013-19-23 Rev: 2: STC description update (including reference to STC 10064496 Rev 3) R...</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Installation in accordance with Rockwell Collins AML RCA-0569, Revision( ), issued November 13, 2017. (Note: This change/repair is subject to the installation of the new IFE equipment. It is recommended that the installation of this modification be performed under the guidance of the manufacturer.)</td>
<td>- For model specific limitations, see AML: RCA-0569, Rev. ( ) Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/01/2024</td>
</tr>
<tr>
<td>10069406</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>C900GT</td>
<td>Modification of Rockwell Collins, Inc. Pro Line Fusion® software and hardware.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/approval.</td>
<td>Installation in accordance with Rockwell Collins AML RCA-0569, Revision( ), issued November 13, 2017. (Note: This change/repair is subject to the installation of the new IFE equipment. It is recommended that the installation of this modification be performed under the guidance of the manufacturer.)</td>
<td>- For model specific limitations, see AML: RCA-0569, Rev. ( ) Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10069414</td>
<td>BOMBARDIER Inc.</td>
<td>REV. 1</td>
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<td>BD-700-2A12</td>
<td>Installation of Mounts for Tablet from Factor Portable Electronic Flight Bag (EFB) Devices Rev. 1: Update of the STC holder address.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Modification Data Summary, Bombardier Inc. document No. RAZ-BM700-167. Revision date: 23 November 2018</td>
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<td>10069416</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 2</td>
<td>FAA ST04319AT-D</td>
<td>SEE FAA AML GVI (G650)</td>
<td>Installation of Satcom Direct Router - Gateway(SDR-D) System: Revision 1: Minor edits on the referenced documents (Rev. 2: Adds GVI-G650 and GVI-660 to the AML).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- GVI-G650 and GVI-G650 Certification Plan GCT3422C001 Revision B - Summary Compliance Report GCT3422C002 Revision A for GVI-G550 - Summary Compliance Report GCT3422C002 Revision B for GVI-G660 - EASA Validation Summary Compliance Report GCT3422C005 Revision (-) for GVI-G660 and GVI-G550 - Index LC...</td>
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<td>10069421</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 3</td>
<td>208, 208B</td>
<td>208, 208B</td>
<td>208, 208B Parachute Kit: Revision 1: Transfer from Atlas Air Service AG to Gomolzig Aircraft Services GmbH. Revision 2: Transfer from Gomolzig Aircraft Services GmbH to ACC Columbia Jet Service GmbH. Revision 3: Type in company name</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Master Document List MODC1-2050230-rev_0_20190615 - SPOH: C208_2050230_rev_0_20180615...</td>
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<td>ELBE FLUGZEUGWERK E GmbH</td>
<td>A319-133</td>
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<td>Structural Modification for Antenna and Window Plug Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- Top Engineering Order, Ref. EF-W EO-0101-001, Rev. A - Airworthiness Limitation Section (ALS) / Part 2 Airworthiness Limitation Items, Ref. A319-ALI-Pegasus/rev1-1: As defined in the AFMS, Ref A319-AFM-PEGASUS (Limitations Section). As defined in the Airworthiness Limitation Section (ALS) / Part 2 Airworthiness Limitation Items, Ref. A319-ALI-Pegasus (Section 3). Prior to installation of the...</td>
<td>- Installation of STC 10069424 (corresponding to EASA project 010046528) - As defined in the AFMS, Ref A319-AFM-Pegasus (Limitations Section). As defined in the Airworthiness Limitation Section (ALS) / Part 2 Airworthiness Limitation Items, Ref. A319-ALI-Pegasus (Section 3). Prior to installation of the...</td>
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<td>LUFTHANSA TECHNICK AG</td>
<td>A319-133</td>
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<td>A319 Special Mission and Cabin Modification: Revision 1: Reduction of minimum cabin crew.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: CS25.7(c)(b), CS25.7(b)(d), CS25.7(b), CS25.7(c)(b), CS25.7(b)(d), CS25.7(b)(c)(d), CS25.7(c)(b)(d), CS25.7(b)(c)(d), CS25.7(b)(c)...</td>
<td>- Supplemental Type Design Document, Ref. ST060-170M, Issue 02/Revision 0-Instruction for Continued Airworthiness, Ref. A-00-1762-IDA, Issue 02/Revision 0-4Airline Flight Manual Supplement, Ref. A-00-1762-AFM, Issue 0/Revision 0-1Flight Crew Data, Ref. A-00-1762-FCD, Issue 01 Revision 0-Master...</td>
<td>- Non-commercial operations only. Installation of STC 10069423 (corresponding to EASA project 0100494529) Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>LUTHANSA TECHNIX AG</td>
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<td>Replacement of Window Plugs by Mission Windows</td>
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<td>GARMIN INTERNATIONAL, Inc.</td>
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<td>SEE FAA AML</td>
<td>Installation of Garmin GFC 500 Autopilot with Electronic Stability</td>
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<td>and Protection. Revision 1 adds new Part 23 aircraft models. Revision 2</td>
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<td>adds new models to the AML. Revision 3 adds new models to the AML.</td>
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<td>AIRBUS HELICOPTERS, Inc.</td>
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<td>Installation of AEM 600W PA Speaker System on EC 130 T2</td>
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<td>BRITISH AIRWAYS PLC</td>
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<td>747-400</td>
<td>Installation of Low Frequency Underwater Locator Beacon</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>10069470</td>
<td>505</td>
<td>Installation of High Skid Landing Gear and Crew Steps in accordance with</td>
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<td>Master Drawing List AA-17071 in relation to FAA STC SR112360G.</td>
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**Certification basis**

- Top Engineering Order, Ref. EFV-EU-011-001, Rev. A: Master Minimum Equipment List, Rev. A319- MME-PEGASUS02, Rev. 01 or later revisions of the above listed document(s) approved/accepted under the EASA system.

Installation of STCs: 10069423 and 10069444. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

Installation of STCs: 10069423 and 10069444. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

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EASA STC no.: FAA STC SR09506RC

Certification basis: FAA STC SR09506RC

See FAA AML

List of EASA Supplemental Type Certificates
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<th>Certification Basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>10069514</td>
<td>BABCOCK MISSION CRITICAL SERVICESDESIGN AND COMPLETIONS LIMITED</td>
<td>STC Holder</td>
<td>05/04/2019</td>
<td>EC135 P1</td>
<td>Single Pilot VFR Conversion.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-27 Amendment 3 further following paragraphs 27.1 (a), 27.301(a), (b), 27.303; 27.305 (4); 27.307(a); 27.312(a); 27.337 (4)…</td>
<td>Rotorcraft Flight Manual Supplement FMS-160-002 Issue Instructions for Continued Airworthiness ICA/160-012 Issue 1Modification Leaflet &amp; Certificate of Design Babcock/MOD/160 Issue 10Design Report OR/ 160-011 Issue for later revisions of the above listed document(s) approved/ accepted/understand the EASA system.</td>
<td>This STC covers only EC135 P1(C) helicopters equipped with analogue instrumentation. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/04/2019</td>
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<td>10069522</td>
<td>COMMUTER AIR TECHNOLOGY Inc.</td>
<td>FAA</td>
<td>04/04/2019</td>
<td>B300, B300C</td>
<td>Operating Weight Increase. Operating Weight Increase, increase gross weight to 7,484 kg (16,500 lb)MTOM.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: as per FAA STC(ASA34342CH) This certificate/approval involves change to the requirements for environmental protection or all…</td>
<td>Operating Weight Increase in accordance with Y/ETS Master Data List(102575), at Revision N, dated January 23, 2017; For the configuration G1, Airplane Flight Manual Supplement for Operating Weight Increase Beechcraft King Air Model B300 and B300C,Commuter Air Technology document 300V7001, at revision…</td>
<td>If the holder agrees to permit another entity to use this approval to modify the product, the holder must give the other entity written evidence of their permission. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previous…</td>
<td>04/04/2019</td>
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<td>10069522</td>
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<td>10069523</td>
<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>REV. 1</td>
<td>A319-112</td>
<td>AFMS MTOW Reduction Rev 1: Transfer of STC from GVH Aerospace Ltd - United Kingdom, to GVHAerospace (Ireland) Limited</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and any change/repair and this certificate/approval.</td>
<td>GVH Aerospace Airplane Flight Manual Supplement No. GVH-6286-63-301, Issue 4, dated 2012; or later FAA approved revision FAA approved for the following paragraphs: 27.1(a), 27.301(a), 27.303; 27.305 (4); 27.307(a); 27.312(a); 27.337 (4)…</td>
<td>Prior to installation of this design change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/01/2021</td>
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<td>10069527</td>
<td>AIR COMM CORPORATION L.L.C</td>
<td>FAA STC</td>
<td>S-76A, S76B, S-76C</td>
<td>SH5778SW</td>
<td>Addition of Electrically Driven Vapor Cycle, Freon or Refrigerant R-134a, Air Conditioning System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SH5778SW. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and any change/repair and this certificate/approval.</td>
<td>Drawing List DL-85-02-100, Revision AC, dated 15 April 2012, or later FAA approved revision FAA approved Rotorcraft Flight Manual Supplement, Document No. CR-478-9, Rev. A, dated 4 March 1997, or later FAA approved revision Instructions for Continued Airworthiness are contained in the Maintenance Manual…</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10069534</td>
<td>POLYTECH AERO SUPPORT INC.</td>
<td>TCCA</td>
<td>A340-642, A340-312, A340-313, A340-541, A340-211, A340-212, A340-311</td>
<td>A340 Improved Air Filter Element - IDPRefer to MS-17-006 Modification Summary</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product and any change/repair and this certificate/approval.</td>
<td>MS-17-006 A340 Improved Air Filter Element Subtitle ModificationSummary Rev 1: EB 21-17-006-1 Engineering Bulletin Rev 0 for later revisions of the above listed document(s) approved/accepted/understand the EASA system in accordance with the Technical Implementation Procedures of EUL Canada Bilateral Agreement…</td>
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<td>MICRO AERODYNAMICS, INC.</td>
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<td>SA00170SE</td>
<td>Installation of vortex generators on the wings and tail surfaces</td>
<td>- Following modifications, all design changes and/ or repairs will introduce no adverse effect upon the airworthiness of the product.</td>
<td>- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>25/03/2020</td>
<td>Aircraft structure reinforcement for ARINC 791 Large Antenna Installations on Airbus A330-202/-203/-243</td>
<td>- Improper installation/installataion may result in the structure exceeding strength requirements.</td>
<td>- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>07/10/2022</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following paragraph(s):</td>
<td>- Improper installation/installataion may result in the structure exceeding strength requirements.</td>
<td>- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>21/06/2024</td>
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<td>- Improper installation/installataion may result in the structure exceeding strength requirements.</td>
<td>- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>- Improper installation/installataion may result in the structure exceeding strength requirements.</td>
<td>- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>Lavatory modifications on Boeing 737-700-800/800/800ER/800ER</td>
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<td>Lavatory modifications on Boeing 737-700-800/800/800ER/800ER</td>
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<td>AVOX SYSTEMS Inc.</td>
<td>FAA STC</td>
<td>FAA STC</td>
<td>ST0569NY</td>
<td>AVOX LavOx A330 Installation of AVOX LavOx Passenger Lavatory Oxygen System</td>
<td>FAA STC</td>
<td>AVOX LavOx A330 Installation of AVOX LavOx Passenger Lavatory Oxygen System</td>
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<td>AEROSPACE DESIGN &amp; COMPLIANCE LLC</td>
<td>FAA STC</td>
<td>FAA STC</td>
<td>ST0401NY</td>
<td>Garmin GTX 300 Installation of the Garmin GTX 300 Automatic Dependent Surveillance - Broadcast (ADS-B) Out System</td>
<td>FAA STC</td>
<td>Installation of the Garmin GTX 300</td>
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<td>1006586</td>
<td>S-TEC CORPORATION</td>
<td>FAA STD SA09751DS</td>
<td>425, 421A, 421B, 421C, 414, 414A</td>
<td>Installation of an S-TEC 01326 Digital Flight Control System (DFCS).</td>
<td>The Certification Basis for the original product as amended by the following: additional or alternative airworthiness requirements: CS 23.1310 Internal Avionics, CS 23.1308 Avionics, CS 23.1309 Avionics, CS 23.2540 Avionics, 773A. The requirements for environmental protection and the associated/standardized noise and/or emissions levels of the or...</td>
<td>1. Master Data List No. ST-966-MDL-0001, Revision A, dated 24 September 2018 or later FAA approved revision 2. Airplane Flight Manual Supplement, ST-966-AMPS-0001, Rev., dated 26 September 2018, or later FAA approved revision 3. Instructions for Continued Airworthiness, ST-966-ICOA-0001, Revision Dat...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/04/2019</td>
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<td>1006594</td>
<td>ARMSTRONG AEROSPACE Inc dba ASTRONICS ARMSTRONG AEROSPACE</td>
<td>FAA STC ST04377CH</td>
<td>737-800-700/800-900-900E</td>
<td>Installation of a Ballard webCS system</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC ST04377CH. The requirements for environmental protection and the associated/standardized noise and/or emissions levels of the or...</td>
<td>Astronics Armstrong Aerospace Master Data List, Document No. DC104-9841-00, Revision B, dated 11 February 2019, or later FAA approved revision. FAA Approved Flight Manual Supplement as identified in Astronics Armstrong Aerospace Master Data List, Document No. DC104-9841-00, Revision B, dated 11 February 2019.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>1006604</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td>525, 525A, 525B, 525C</td>
<td>ADS-B OUT GNSS source modification for Flight 2 Replacement Rockwell Collins GPS-400S by Garmin GTX-7400 &amp; GTX-7400CFR installation to transponder 2 (Rockwell Collins TDR-94).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/standardized noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>- Modification in accordance with MDL 2019-1-16 Revision (Rand Engineering Order 2019-1-16 Revision IR). Installation in accordance with Installation Procedure 2019-1-16 Revision IR. - Installation approved in accordance with Airplane Flight Manual Supplement, AMF052019-1-8 Revision IR. Maintenance and/or...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>1006607</td>
<td>PATS AIRCRAFT L.L.C dba PATS AIRCRAFT SYSTEMS/aivia ALOFT AERONAUTIC S</td>
<td>REV. 1 FAA ST03741NY</td>
<td>A321-231/232 A321-211/212-213 A321-111/112-120 A319-111/112-130 A319-111-112-113/114-115</td>
<td>KA SATCOM Antenna System Installation. The STC consists in the installation of both internal radome and antenna along with the interlakena power supply unit and satellite modem. Through this amendment to STC ST03741NY for ViaSat KA SATCOM AntennaSystem, installation on Airbus A319 Series Aircraft...</td>
<td>The Certification Basis for the original product as amended by the following: additional or alternative airworthiness requirements: Equivalent Safety Finding(s): CRI B-4301-007 - Vibration and buffeting compliance criteria for large external antenna installations the following paragraph(s) at a later amen...</td>
<td>- MDL ViaSat document No. 1272199 Revision 012, ICA ViaSat document No. 1277470 Revision 005, EWIS ICA ViaSat document No. 1277468 Revision 002, AMPS ViaSat document No. 1277466 Revision 001, AMPS (A319) ViaSat document No. 1283754 Revision 001 for later revisions of the above listed document(s) a...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>1006610</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC SR00917DE</td>
<td>S-90A</td>
<td>Installation of Concorde VRLA Battery</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SR00917DE. The requirements for environmental protection and the associated/standardized noise and/or emissions levels of the or...</td>
<td>Document No. S-90A, Master Drawing List, Skrynsky S-90A Battery Replacement, Revision N/C, dated 19 September 2018, FAA approved revision 2018, or later FAA approved revision. FAA approved 2018, or later FAA approved revision. Instructions for Continued Airworthiness (ICA), Document No. S-90A-ICA Revision IR, dated 19 September 2018, FAA accepted 2 Nov...</td>
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<td>PMV ENGINEERING</td>
<td>Installation of a portable EFB arm mount on A350</td>
<td>- FAA STC SD0282LA DHC-6 SERIES 300 DHC-6-300 Max Take-off Weight Increase up to 6305 kg (14,000 lbs). Increase of Landing Weight up to 6260 kg (13,800 lbs). Installation of PT9A-3A engines rated at 750 SHP and Hartzell HC-D4N-3A / D9515 4-bladed propellers as defined in IKHANA Aircraft Services Installation Drawing List ER218-3000, Rev...</td>
<td>Airplane Flight Manual Supplement (AFMS) HGLD-63-3-A, Rev...</td>
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<td>IKHANA GROUP, LLC</td>
<td>Introduction of Underwater Locator Device</td>
<td>- FAA STC SA02682LA DHC-6 SERIES 300 DHC-6-300 Max Take-off Weight Increase up to 6305 kg (14,000 lbs). Increase of Landing Weight up to 6260 kg (13,800 lbs). Installation of PT9A-3A engines rated at 750 SHP and Hartzell HC-D4N-3A / D9515 4-bladed propellers as defined in IKHANA Aircraft Services Installation Drawing List ER218-3000, Rev...</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>Introduction of Underwater Locator Device</td>
<td>- FAA STC SA02682LA DHC-6 SERIES 300 DHC-6-300 Max Take-off Weight Increase up to 6305 kg (14,000 lbs). Increase of Landing Weight up to 6260 kg (13,800 lbs). Installation of PT9A-3A engines rated at 750 SHP and Hartzell HC-D4N-3A / D9515 4-bladed propellers as defined in IKHANA Aircraft Services Installation Drawing List ER218-3000, Rev...</td>
<td>Airplane Flight Manual Supplement (AFMS) HGLD-63-3-A, Rev...</td>
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<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>Replacement of the HT1000 MCDU (Multi-Function Control Display Unit) with the HT9100 MCDU</td>
<td>- FAA STC ST0443AT ATR 72-211, ATR 72-212 ATR 72-201, ATR 72-202 ATR 72-101, ATR 72-102 ATR 42-320, ATR 42-500 ATR 42-200, ATR 42-300 Chippewa Aerospace Master Data List No. 42-3461-02-0001, Rev...</td>
<td>Chippewa Aerospace Master Data List No. 42-3461-02-0001, Rev...</td>
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<td>DELTA ENGINEERING</td>
<td>Installation of Aviation Technical Services CargoTailTM.</td>
<td>- FAA STC ST0461NY ATR 72-211, ATR 72-212 ATR 72-201, ATR 72-202 ATR 72-101, ATR 72-102 ATR 42-320, ATR 42-500 ATR 42-200, ATR 42-300 Chippewa Aerospace Master Data List No. 42-3461-02-0001, Rev...</td>
<td>Chippewa Aerospace Master Data List No. 42-3461-02-0001, Rev...</td>
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List of EASA Supplemental Type Certificates

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<th>EASA STC No</th>
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<tr>
<td>10069621</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST04314AT-D</td>
<td>GIV, GV</td>
<td>Installation of Honeywell Laser/IV RUU</td>
<td>Certification Basis for the original product remains applicable with certificate approval, except where amended by additional or later amendments if indicated on FAA STC ST04314AT-D. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/FAA STC 10015732(EASA.IM.A.S.01754).</td>
<td>Gulfstream Aerospace Corporation Index List CES11139001, Revision A dated 3 October 2018, or later FAA approved revision FAA accepted Instructions for Continued Airworthiness - G-IV Model: CES11139001, Revision A dated 29 July 2018, or later FAA approved revision - G-IV Model: CE411134001, dated 28 Aug.</td>
<td>21/06/2019</td>
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<td>10069622</td>
<td>BOMBARDIER Inc.</td>
<td>REV. 1</td>
<td>TCCA SA15-132 ISSUE 1</td>
<td>BD-700-2A12</td>
<td>Installation, Aircraft Completion Modification Rev. 1: Update of the STC holder address.</td>
<td>Certification Basis for the original product as amended by the following paragraphs: 19. The requirements for environmental protection and the associated certified noise and/or emissions levels.</td>
<td>Configuration Definition Record, Bombardier Inc. document No. RAZ-BM700-16A Revision D dated 10 April 2019(Supplemental Airplane Flight Manual, Bombardier Inc. document No. DCA-7500-001 Revision Basic dated 9 November 2019 Instructions for Continued Airworthiness, Bombardier Inc. document No. GCD0000X7500.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/04/2023</td>
<td>Active</td>
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<td>10069624</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA ST04072NY</td>
<td>777-300ER</td>
<td>Installation of a Cabin Interior.</td>
<td>Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-25 Amendment 19. The requirements for environmental protection and the associated certified noise and/or emissions levels.</td>
<td>Delta Engineering Master Data List 0143-19482-2455 Revision B dated January 2019 or later FAA approved revisions Instructions for Continued Airworthiness as listed on the Master Data List.</td>
<td>Master Data List Configuration 1 STC 10014897 (ST02535NY) Issued/For Master Data List Configuration 1 Airframe Industries Service Bulletin DFSB-8777-18-045-002 and SRTService Bulletin CN-01888-CCE are required prior to installation of this STC For Master Data List Configuration 1 Air</td>
<td>15/04/2019</td>
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<td>10069627</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 1</td>
<td>KODIAK 100</td>
<td>Installation of MTV-27(-) propeller on the Gulfstream 100. This modification removes the existing reversing and full feathering/Hartzell propeller with the appropriate spinner and replaces it with the blade MTV-27-1-E-C-F-R-P(7 CPR240-55) propeller and spinner. The existing governor and engine co.</td>
<td>Certification Basis for the original product as amended by the following paragraphs: 19. The requirements for environmental protection and the associated certified noise and/or emissions levels.</td>
<td>According to MT-Propeller Master Data List No. E-3113 Rev. dated 03/02/2021. To be operated in accordance with MT-Propeller AFMS Document No. E-3115 Rev. 0 dated 04/04/2019. To be installed Instruction Doc. No. E-3116 Rev. 3 dated 06/02/2020. To be maintained in accordance with MT-P.</td>
<td>Note: This STC is compatible with the EASA STC 10061979 (Gulfstream II or III) and is not approved for an AHRS replacement. - Application of the -03 Configuration is limited to serial numbers 501-584 without Gulfstream ASC 82. For Gulfstream G-IV aircraft: Application of the -01 Configuration is only approved to replace LASERREF or III and is not approved for an AHRS replacement.- Application of the -03 Configuration is...</td>
<td>04/02/2021</td>
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<td>10069642</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>727-400</td>
<td>Cargos of Horses and Other Animals Revision 1: ‘Limitations/Conditions’ updated to remove reference EASASTC 10015780.</td>
<td>Certification Basis for the original product as amended by the following paragraphs: 19. The requirements for environmental protection and the associated certified noise and/or emissions levels.</td>
<td>- Flight Manual Supplement S21.FMS-0154 Issue 3- Master Data List S21.LM-1416 Issue 3 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The Following EASA STCs must have been previously embodied or embodied concurrently with this design change: EASA STC 1001572 (Gulfstream IV A.S.01754) and EASA STC 10031361. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/05/2019</td>
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<td>Date of Issue</td>
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<td>Certification basis</td>
<td>Approval of Antenna Structural Provision</td>
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<td>17/04/2019</td>
<td>10069670</td>
<td>Active</td>
<td>This STC may only be installed in conjunction with ACSS NXIG-900 GPS Receiver installation i.e. EASA STC No. 10069624 FAA STC No. ST02832LA-Prior to installation of this change/repair it must be determined that the aircraft's previous modifications of this type are no adverse effect upon the airworthiness of the product.</td>
<td>Approval of Antenna Structural Provision under Morson Projects Ltd Modification ref MP0030A.</td>
<td>Approval of Antenna Structural Provision under Morson Projects Ltd Modification ref MP0030A and Service Bulletin ref SB.MP51-036. Instructions for continued airworthiness in accordance with Morson Projects Ltd ICA ref MPL-C-04-AW0141 Iss 2, dated April 2019. There is no Flight Manual change associated with this...</td>
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<td>1006863</td>
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<td>Prior to installation of this change/repair it must be determined that the aircraft's previous modifications of this type are no adverse effect upon the airworthiness of the product.</td>
<td>Approval of Antenna Structural Provision under Morson Projects Ltd Modification ref MP0030A.</td>
<td>Approval of Antenna Structural Provision under Morson Projects Ltd Modification ref MP0030A and Service Bulletin ref SB.MP51-036. Instructions for continued airworthiness in accordance with Morson Projects Ltd ICA ref MPL-C-04-AW0141 Iss 2, dated April 2019. There is no Flight Manual change associated with this...</td>
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<td>23/04/2019</td>
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<td>Active</td>
<td>This approval is subject to the limitations specified in TCCA ADCF-2012-13 in accordance with AVAC Alternative Means of Compliance (AMOC). Alleron No. AARDG 2019/ A13. Prior to installation of this change/repair it must be determined that the aircraft's previous modifications of this type are no adverse effect upon the airworthiness of the product.</td>
<td>Replacement of the Passenger Door Piano Hinge</td>
<td>Approval of Antenna Structural Provision under Morson Projects Ltd Modification ref MP0030A and Service Bulletin ref SB.MP51-036. Instructions for continued airworthiness in accordance with Morson Projects Ltd ICA ref MPL-C-04-AW0141 Iss 2, dated April 2019. There is no Flight Manual change associated with this...</td>
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<td>1006965</td>
<td>Active</td>
<td>Prior to installation of this design change it must be determined that the aircraft's previous modifications of this type are no adverse effect upon the airworthiness of the product.</td>
<td>Modification and replacement of existing Rockwell Collins dual Attitude Heading Reference System (AHRS) AH5-558 with Rockwell Collins dual (AHRS) AH5-3000A.</td>
<td>Approval of Antenna Structural Provision under Morson Projects Ltd Modification ref MP0030A and Service Bulletin ref SB.MP51-036. Instructions for continued airworthiness in accordance with Morson Projects Ltd ICA ref MPL-C-04-AW0141 Iss 2, dated April 2019. There is no Flight Manual change associated with this...</td>
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<td>27/01/2020</td>
<td>1006967</td>
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<td>Prior to installation of this change/repair it must be determined that the aircraft's previous modifications of this type are no adverse effect upon the airworthiness of the product.</td>
<td>Covering and recovering of sailplanes and powered sailplanes with &quot;Coconut&quot; polyester cover.</td>
<td>Approval of Antenna Structural Provision under Morson Projects Ltd Modification ref MP0030A and Service Bulletin ref SB.MP51-036. Instructions for continued airworthiness in accordance with Morson Projects Ltd ICA ref MPL-C-04-AW0141 Iss 2, dated April 2019. There is no Flight Manual change associated with this...</td>
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<td>23/04/2019</td>
<td>1006969</td>
<td>Active</td>
<td>Prior to installation of this change/repair it must be determined that the aircraft's previous modifications of this type are no adverse effect upon the airworthiness of the product.</td>
<td>Installation of Emergency Float System</td>
<td>Approval of Antenna Structural Provision under Morson Projects Ltd Modification ref MP0030A and Service Bulletin ref SB.MP51-036. Instructions for continued airworthiness in accordance with Morson Projects Ltd ICA ref MPL-C-04-AW0141 Iss 2, dated April 2019. There is no Flight Manual change associated with this...</td>
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<td>Model</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>CHIRPEWA AEROSPACE, Inc.</td>
<td>ATR 72-211, ATR 72-212, ATR 72-201, ATR 72-202, ATR 72-101, ATR 72-102, ATR 42-500, ATR 42-300, ATR 42-320, ATR 42-200</td>
<td>Replacement and upgrade of HT1000 NRP (Nav Processor Unit) and the HT9100NRP.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST0437AT. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>1. Chippewa Aerospace Master Drawing List (MDL) No. 42-3461-1, dated 2 August 2018, or later FAA approved revision.</td>
<td>None</td>
<td>12/02/2019</td>
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<td>10069710</td>
<td>AERO-DIENST GMBH</td>
<td>SA 55</td>
<td>5S5B 55</td>
<td>Replacement of the GA100 CVR by the L3 FA2100 CVR.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative enivronmental requirements – Special Condition (s): Non-rechargeable Lithium Battery. Installation of the transponder to a P/N BB1569. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change(s) or repair will not introduce adverse effect upon the airworthiness of the product.</td>
<td>According to Aero-Dienst Master Drawing List Document No. 2031-LJ.DC-101, Rev.00; To be operated in accordance with FAA Airworthiness Data Base Document No. 2031-LJ.DC-106, Rev.00; To be maintained in accordance with FAA Airworthiness Data Base Document No. 2031-LJ.DC-105 Rev.00 or later revisions of the above listed document(s) approved/accepted.</td>
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<td>10069712</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST069120</td>
<td>GULFSTREAM 200/200i GALAXY</td>
<td>Installation of the Future Air Navigation System (FANS) 1/A+-, Remote Oceanic Controller Data Link Communications (ROCDLC), Automatic Dependent Surveillance - Contract (ADS-C), and CPDLC, Department of Defense (CPDLC-DCL) applications through replacement of the existing Honeywell/Avionics Flight Information Processor (FIP) with the Iridium SATCOM System (FANS-1/A+).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST069120. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>1. Master Drawing List E-TM-17-0002, Rev. 8, dated 13 February 2019; or later FAA approved revision 2 or later FAA approved revision 3. Airplane Flight Manual Supplement E-17-0002.</td>
<td>A prerequisite to this STC is the installation of Iridium SATCOM System (FANS-1/A+) on the aircraft. This STC is specific to the following aircraft(s): None</td>
<td>07/12/2021</td>
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<td>10069712</td>
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<tr>
<td>10069713</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST04395AT-D</td>
<td>GVI (G650)</td>
<td>Installation of the Iridium SATCOM System (FANS-1/A+) &lt; Over Iridium.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST04395AT-D. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>1. Gulfstream Aerospace Corporation Index List GDL100000028, Revision ( ), dated 23 July 2018, or later FAA approved revision 2. Airplane Flight Manual Supplement, CE91 203M001, Revision ( ), dated 23 July 2018, or later FAA approved revision 3. Airplane Flight Manual Supplement, CE91 203M002, Revision ( )</td>
<td>1. Iridium SATCOM System requires FANS-1/A+ to operate; therefore FANS-1/A+ must be installed prior to or concurrently with this STC. This STC does not provide the necessary “Operational Approval” for the use of the FANS 1/A+ functions. This STC is specific to the following aircraft(s): None</td>
<td>24/04/2019</td>
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<td>10069725</td>
<td>URBE AERO GmbH</td>
<td>200</td>
<td>TDR94D transponder upgrade ADS-B</td>
<td>Installation limited to Beech King Air 200 aircraft with serial number BB1569 for ADS-B out compliance with CS-AACS.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Installation in accordance with Engineering Order Doc. D6E1F01, Project No. 147, Issue 01-. Operation in accordance with Supplement to the Airplane Flight Manual, Doc. No. AFM-147-01, Date of Issue 29-Mar-2019-. Maintenance and Instructions for Continued Airworthiness in accordance with Doc. No. AMM.</td>
<td>None</td>
<td>25/04/2019</td>
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<td>SABENA TECHNIFS BGC</td>
<td>Cabint Reconfiguration to 295 Pax with IFR System on A330-300. Revision 1 consisted of a full cabin reconfiguration to reach 295 Pax(30B/C, 21PY/C, 244/Y, 14 E) and a newery of the 2If H System. Revision 2 with extension to pre-mod configuration as recorded in Master Data List. Revision...</td>
<td>- Instructions for continued airworthiness 40121-02-A-ICA-050 - SSD Change Document 40121-02-A-OSDCD-R00 - Master Data Document 40121-02-AMDL-R01 - Aircraft Flight Manual Supplement no. 40121-02-A-AMFNS-R00</td>
<td>- Continued airworthiness 40121-02-A-ICA-050.</td>
<td>14/02/2020</td>
<td>Active</td>
<td>10069731</td>
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<tr>
<td>PRATT &amp; WHITNEY</td>
<td>Aircraft Configuration change to the existing cabin arrangement to increase overall seating capacity from Zero Occupancy Airbus Model 100 Pax to 196 Pax for A319/VOLOTEAirbus Revision 1 - Integration of the models: A319-111/A319-113/A319-114/A319-115/A319-131/A319-132/A319-133 in a...</td>
<td>- Certification Summary Report MOD 01007TAS-AUA-180034C issue 0.0 prior to installation of this change/repair and any other pre-installed change/repair and any other pre-installed change/repair.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelation of this change/repair and any other pre-installed change/repair and any other pre-installed change/repair and any other pre-installed change/repair.</td>
<td>10/12/2020</td>
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<td>J &amp; C AEROSPACE DESIGN AND CERTIFICATI ON SERVICES EEU LTd.</td>
<td>Placards &amp; markings MMEL Supplement this Standalone MMEL supplement is used for the management of missing unreadable placards and markings. Rev. 1: applicable list extended other models.</td>
<td>- Certification Summary Report MOD 01007TAS-AUA-180034C issue 0.0 prior to installation of this change/repair and any other pre-installed change/repair and any other pre-installed change/repair.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelation of this change/repair and any other pre-installed change/repair and any other pre-installed change/repair...</td>
<td>30/04/2019</td>
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<td>PRATT &amp; WHITNEY</td>
<td>Installation of a Pratt &amp; Whitney enhanced Flight Data Acquisition Storage and Transmission (eFAST) system.</td>
<td>- Certification Summary Report MOD 01007TAS-AUA-180034C issue 0.0 prior to installation of this change/repair and any other pre-installed change/repair and any other pre-installed change/repair...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelation of this change/repair and any other pre-installed change/repair and any other pre-installed change/repair...</td>
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<td>PRATT &amp; WHITNEY</td>
<td>Installation of a Pratt &amp; Whitney enhanced Flight Data Acquisition Storage and Transmission (eFAST) system.</td>
<td>- Certification Summary Report MOD 01007TAS-AUA-180034C issue 0.0 prior to installation of this change/repair and any other pre-installed change/repair and any other pre-installed change/repair...</td>
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<tr>
<td>DOA21 LIMITED</td>
<td>REV. 1</td>
<td>A330-243</td>
<td>Introduction of a Cabin VIP Interf: Rev. 1</td>
<td>Transfer of STCholder role to DOA21 Ltd.</td>
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<td>ACC AIRCRAFT CABIN COMPONENTS GmbH</td>
<td>REV. 1</td>
<td>PA-28-180 (ARCHER)</td>
<td>Modification of Piper PA-28-180 S/N 28-7305200 rear seats to accommodate children</td>
<td></td>
<td>02/05/2019</td>
<td>Active</td>
<td>10067802</td>
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<td>ROSEN SYSTEMS L.L.C.</td>
<td>REV. 1</td>
<td>FAA STC SR00840DE</td>
<td>Installation of the Rosen Survivor System into the Robinson R22, R44 and R66</td>
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<td>22/07/2019</td>
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<td>FOKKER SERVICES B.V.</td>
<td>REV. 3</td>
<td>747-400F 747-400</td>
<td>Activated GNSS Stand-Alone provisions for ADS-B Out. This STC activates the provisions introduced with EASA STC 10068342a-747-400F and introduces activated provisions on 747-400. (Revision 1, Revision 2 and Revision 3 were issued to correct information on the certificate)</td>
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<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERSERVING ES ING. WALTER STARZACHER GmG</td>
<td>REV. 1</td>
<td>AML STC for ELT Installation/EASA AML STC for installation of portable and survival type ELT’s on various aircraft types. The following equipment are approved with - STC for FAA: KANNAD/ GROLIA PN S18020502-1, S1823502-2, S1838501-1, TECHTEST PN H 500-32-2Y-G, 500-32-2Y-H, EXAIL PN1028768-1</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s) CRI F-01.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/10/2023</td>
<td>Active</td>
<td>10067972</td>
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</table>
Deep Snow Kit Installation, composed of two Bear Paws at the aft end of the skis and a trampolined running between the paws, along with the associated clamps and hardware.

EASA Validation of Transport Canada Supplemental Type Certificate (STC) SA19-13. The Certification Basis for the original product as amended by applicable STC/SA19-13 Installation Data, dated 23 February 2019. The Certification Basis for the original product remains applicable to this certificate/STC.

SAI-19-13 Installation, Fire Extinguishers with Halocarbon Clean Extinguishing Agent. Modification must be in accordance with the following approved/accepted under the above listed document(s).

Modification is made in accordance with the following FAA STC: BD-100-1A10 aircraft model: Modification Data Summary 476288, dated 15 November 2017: FAA STC 2004-015-002, Rev. No. 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

Modification is made in accordance with the following FAA STC: BD-100-1A10 aircraft model: Modification Data Summary 476288, dated 15 November 2017: FAA STC 2004-015-002, Rev. No. 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

Modification is made in accordance with the following FAA STC: BD-100-1A10 aircraft model: Modification Data Summary 476288, dated 15 November 2017: FAA STC 2004-015-002, Rev. No. 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

Modification is made in accordance with the following FAA STC: BD-100-1A10 aircraft model: Modification Data Summary 476288, dated 15 November 2017: FAA STC 2004-015-002, Rev. No. 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

Modification is made in accordance with the following FAA STC: BD-100-1A10 aircraft model: Modification Data Summary 476288, dated 15 November 2017: FAA STC 2004-015-002, Rev. No. 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

Modification is made in accordance with the following FAA STC: BD-100-1A10 aircraft model: Modification Data Summary 476288, dated 15 November 2017: FAA STC 2004-015-002, Rev. No. 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

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Modification is made in accordance with the following FAA STC: BD-100-1A10 aircraft model: Modification Data Summary 476288, dated 15 November 2017: FAA STC 2004-015-002, Rev. No. 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

Modification is made in accordance with the following FAA STC: BD-100-1A10 aircraft model: Modification Data Summary 476288, dated 15 November 2017: FAA STC 2004-015-002, Rev. No. 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

Modification is made in accordance with the following FAA STC: BD-100-1A10 aircraft model: Modification Data Summary 476288, dated 15 November 2017: FAA STC 2004-015-002, Rev. No. 1. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.
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<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tr>
<td>10069829</td>
<td>STC TWENTY ONE LIMITED</td>
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<td>CAA STC RESPONSIBILITY</td>
<td>BAE 146 SERIES 100</td>
<td>Approval of P-RNAV and UNAV RNP Approach</td>
<td>- STC Twenty One Ltd Flight Manual Supplement No. S21-045-017 Issue 1- STC Twenty One Ltd Maintenance Manual / AIP Supp</td>
<td>Restricted to MSN E114-STC Twenty One Limited Design Change S21-3A-25-1194 must be embodied prior to embodiment of this Design Change. This Design Change is limited to RFIS-equipped aircraft only. Unless the allocation conditions formulated in Master Minimum Equipment List Supplement No. S21-MMEL-0023...</td>
<td>09/02/2019</td>
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<td>10069831</td>
<td>SAFRAN CABIN INC.</td>
<td>REV. 1</td>
<td>FAA ST01408LA</td>
<td>CL-600-2B19</td>
<td>Installation of Reinforced Cockpit Doors with Flight Deck Access System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>C &amp; D Interiors Controlled Drawing List, 4E11000, Revision B, dated 9/December 2002, or later FAA approved revision.</td>
<td>09/05/2019</td>
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<td>10069838</td>
<td>CERTIFICATION, LLC</td>
<td></td>
<td>FAA ST0269CH</td>
<td>737-800, 737-900ER</td>
<td>Rarotori installation of Honeywell TRA-100B Automatic Dependence Surveillance - Broadcast (ADS-B) Out ATC Transponder System.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:The following paragraph(s) at a later amendment: CS25 Initial issue: 25.29(a)(b); 25.301(a); 25.303; 25.309(a)(b); 25.309(c); 25.451(c) (d); 25.451(d); 25.771(a); 25...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product...</td>
<td>10/05/2019</td>
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<td>10069859</td>
<td>E &amp; D AEROSPACE AIRWORTHINESS</td>
<td></td>
<td></td>
<td>B200</td>
<td>ASPIRE 260 SATCOM INSTALLATION ON B200</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:The following paragraph(s) at a later amendment: CS25 Initial issue: 25.34; 25.34; 25.153; 25.165; 25.173 from FAR 23 at amendment 14/23.573 from FAR 23 at amendment 45(1-4) &gt;23.1301 from FAA...</td>
<td>Change according to Master Document List, 01562-EAS-A-MDL, Rev. C, dated 4 JUN 2019 as described in Supplemental Type Design, 01562-EAS-A-STD-01, Revision C, dated JUN 2019 - Airworthiness Limitations and Maintenance Requirements in accordance with instructions for continued airworthiness supplement, 01562-EAS-A...</td>
<td>18/06/2019</td>
<td>Active</td>
<td>10069859</td>
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<tr>
<td>10069860</td>
<td>SABENA TECHNICS BOD</td>
<td>REV. 3</td>
<td></td>
<td>B200 SUPER KING AIR B200</td>
<td>King Air B200 Monitoring System - Configurations 1, 2, 3 - with revised noise levels</td>
<td>- Installed in accordance with Certification Program, Document Number MOCP-B200-05-001, Issue 17 Dated 10 June 2022 and Master Data List Document Number MDL-0-04-4161 Issue D, Dated 22 June 2022 - Operated in accordance with Aircraft Flight Manual Supplement (AFMS)(r-1-2)(o) Document Number: AFMS-B200-0...</td>
<td>- Configuration: Aircraft build standard is a Beechcraft King Air B200 modified with the following Engine and Propellers-equipment: STCs: STC 10040181 (FAA STC SA02715CH-D) - Pratt &amp; Whitney Canada P76A-St200, STC 10033158 Rev. 8 or later as applicable.</td>
<td>24/06/2022</td>
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<td>Limitations/Conditions</td>
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<td>10069861</td>
<td>UNICAL AERO INC.</td>
<td>FAA STC ST02870LA</td>
<td>737-800ER</td>
<td>Installation of Unical Aero USB-PED Charging System</td>
<td>The Certification Basis for the original product remains applicable thh certificates/approval, except as specified by this STC. The requirements for additional or later amendments if indicated on FAA STC ST02870LA: The requirements for environmental protection and associated certification of noise and/or emissions levels of the or...</td>
<td>1. Master Data List (MDL) 000-008, Revision B, dated 6 December 2017, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined if the...</td>
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<td>10069872</td>
<td>ICELANDAIR EHF</td>
<td>FAA STC ST07007LA</td>
<td>757-300EP</td>
<td>- Icelandair ADF-S document reference</td>
<td>AFC-757-3450-046 Rev B</td>
<td>- Icelandair ICA document reference</td>
<td>FAA2020 Mandate. Rev. 1: This change introduces no adverse effect upon the airworthiness of the product.</td>
<td>16/04/2021</td>
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<td>10069884</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>FAA STC SR02622SE</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of Garmin G500H TXi Display System (Primary Flight Display (PFD), Multi-Function Display (MFD))</td>
<td>The Certification Basis for the original product remains applicable thh certificates/approval, except as specified by this STC. The requirements for environmental protection and associated certification of noise and/or emissions levels of the or...</td>
<td>- Garmin Master Drawing List (MDL) 165-01297-00, Revision B, dated 16 November 2018 or later FAA approved revision. - Airframe Flight Manual Supplement (RFMS) 190-02401-02 Rev. 1, dated 16 November 2018 or later FAA approved revision.</td>
<td>This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List FAA STC SR02622SE dated 5 August 2020. A Garmin GPS Navigator source must be available to support the G500H TXiDisplay System.Rotorcraft that are modified per this STC are limited to VFR performance.</td>
<td>11/11/2020</td>
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<td>10069893</td>
<td>B/E AEROSPAC, Inc.</td>
<td>FAA STC ST04464AT</td>
<td>747-400-400F &amp; 747-800F</td>
<td>Installation of the Rockwell Collins Automatic Dependent Surveillance-Broadcast (ADS-B) Out System</td>
<td>The Certification Basis for the original product remains applicable thh certificates/approval, except as specified by this STC. The requirements for environmental protection and associated certification of noise and/or emissions levels of the or...</td>
<td>- Master Drawing List (MDL) 2-203317-001, Revision B, dated 20 March 2019, or later FAA approved revision. - Airframe Flight Manual Supplement 2-203317-805, Revision IR, dated 15 April 2019, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined if the...</td>
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<td>10069906</td>
<td>ARCHEIDON HOLDINGS, LLC.</td>
<td>FAA STC ST03865AT</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of an Avionica, Inc. avCM Cellular Module (Avionica Module) on the following Avionica QuickAccess Recorder models installed by other STCs: MK-III mini QAR, avRDC QAR, or avRDC MAX QAR. The avCM is intended to provide QAR data transfer over the cellular network when the aircraft...</td>
<td>The Certification Basis for the original product remains applicable thh certificates/approval, except as specified by this STC. The requirements for environmental protection and associated certification of noise and/or emissions levels of the or...</td>
<td>- Configuration 1 3G avCM with battery, Instructions for ContinuingAirworthiness 100600-004, Revision D, dated 10 April 2015.</td>
<td>Prior to installation of this design change it must be determined if the...</td>
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<td>STAR ACQUISITION LLC (Air Transport Technologies, INC)</td>
<td>10069922-1</td>
<td>PA-32R-301T (SARATOGA II TCA)</td>
<td>The Certification Basis for the original product remains applicable: systems/ installation/ the associated certified noise and/or emissions levels of the or changes.</td>
<td>1. Installation instructions as listed on MDL-0065, Revision H, dated 12/01/2024, or later FAA approved revisions 2. The following Airlantane Flight Manual Supplement (rev) (AFMS), or later FAA approved revisions must be carried on the aircraft when the following kits are installed. The following A. 14 Volt System aircraft must be installed via its PMA, but not installed in PA-32R-301T aircraft must have an Oildyne hydraulic pump, Piper P/N...</td>
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<td>KAMAN CORPORATION</td>
<td>10069925-1</td>
<td>RAF STC 10069925-1</td>
<td>The Certification Basis for the original product remains applicable: systems/ installation/ the associated certified noise and/or emissions levels of the or changes.</td>
<td>Prior to installation of this design change it must be determined that the installation between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>ARCHACHEON HOLDINGS, LLC</td>
<td>10069926-1</td>
<td>RAF STC 10069926-1</td>
<td>The Certification Basis for the original product remains applicable: systems/ installation/ the associated certified noise and/or emissions levels of the or changes.</td>
<td>Prior to installation of this design change it must be determined that the installation between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>10069938-1</td>
<td>RAF STC 10069938-1</td>
<td>The Certification Basis for the original product remains applicable: systems/ installation/ the associated certified noise and/or emissions levels of the or changes.</td>
<td>This STC is applicable only on aircraft where the EASA STC 10028830 is embodied. Prior to installation of this change/repair it must be determined that the installation between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>OMNIC ENGINEERING AND MANUFACTURING RING, INC.</td>
<td>10069949-1</td>
<td>RAF STC 10069949-1</td>
<td>The Certification Basis for the original product remains applicable: systems/ installation/ the associated certified noise and/or emissions levels of the or changes.</td>
<td>Prior to installation of this design change it must be determined that the installation between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>ROCKWELL COLLINS, INC.</td>
<td>10069951-1</td>
<td>RAF STC 10069951-1</td>
<td>The Certification Basis for the original product remains applicable: systems/ installation/ the associated certified noise and/or emissions levels of the or changes.</td>
<td>Prior to installation of this design change it must be determined that the installation between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
<th>URBE AERO GmbH</th>
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<td>STC Ref.</td>
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<td>Rev.</td>
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<tr>
<td>Model</td>
<td>525</td>
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<tr>
<td>Description</td>
<td>Change 183, L.3 Lynx NGT-9000D + Installation of a L3 NGT-9000D Mode S Transponder and approval for ADS-BOUT according to CS-ACNS. The Transponder is installed in addition to the existing Garmin GTX330D.</td>
</tr>
<tr>
<td>Certification Basis</td>
<td>CFR Part 23, effective February 1, 1985, as amended by Amendments 23- Through 23-40, and 23-40, CS-ACNS, initial Issue. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate.</td>
</tr>
<tr>
<td>Associated Technical Documentation</td>
<td>- Equipment approval according to design change proposal. Doc: DOE F01 Project No. 183, Issue 01 - Installation in accordance with Engineering Order doc: DOE F 07 Project No. 183, Issue 01 - Operation in accordance with Supplement to the Airplane Flight Manual Doc. No: AFM-183-01 - Maintenance and instr...</td>
</tr>
<tr>
<td>Limitations/Conditions</td>
<td>This STC only applies to installations originally fitted with 2 Garmin GTX330 transponders where all required antenna are already available. The NGT-9000D* is installed on the location of the Garmin GTX330D. This STC does not cover antenna installations or relocations. Prior to installation of this change...</td>
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<td>Date of Issue</td>
<td>21/05/2019</td>
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<th>STC Holder</th>
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<td>Model</td>
<td>DHC-8-103</td>
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<td>Description</td>
<td>EGPSWS Modification (SN-593)/Relocation of EGPSWS Terrain Selection, Terrain Warning, Terrain Inhibit and add steep approach Annunciators/Switches.</td>
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<td>Certification Basis</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
</tr>
<tr>
<td>Limitations/Conditions</td>
<td>- The change is defined with FAEL Master Drawing and Data Item(s) designated 26330000, Rev. IR - Maintenance must be performed in accordance with FAEL Maintenance/Manual Supplement designated 26330100. Rev. (For later revisions of the above listed document(s) approved/ accepted on behalf of EASA in accordance...</td>
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<td>Date of Issue</td>
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<td>REV. 1</td>
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<td>Model</td>
<td>777-200, 777-300ER, 777F</td>
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<td>Certification Basis</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
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<tr>
<td>Limitations/Conditions</td>
<td>Supplemental Type Description Document SD2304-1680M Issue 2.0Description of Change A2601-34 Rev 3/Supplement Document A-00-1810-AFM issue Rev 01/03Instructions for Continued Airworthiness A-00-1810-ICA Issue/Rev 2/0or later revisions of the above listed document(s) approved/accepte...</td>
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<td>Date of Issue</td>
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<td>Rev.</td>
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<td>Model</td>
<td>EC 155 B, EC 155 B1</td>
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<td>Description</td>
<td>Installation of Automatic Deployable ELT.</td>
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<tr>
<td>Certification Basis</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
</tr>
<tr>
<td>Limitations/Conditions</td>
<td>Certification Program CPR-155-201-29-01, Rev. 00Compliance Check List CCL-155-201-29-01, Rev. 00Master Document List MO-155-201-29-01, Rev. 00Flight Manual Supplement FMS-SPAES-0201, Rev. 00Declaration of Compliance Change DCC-155-201-25-01, Rev. 00or later revisions of the above listed document(s)...</td>
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<td>Rev.</td>
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<td>Description</td>
<td>Installation of FiReFly WiFi in AIRBUS A318/A319/A320/A321</td>
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<td>Certification Basis</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
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<td>Date of Issue</td>
<td>27/05/2019</td>
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<td>10070044</td>
<td>AIR ALLIANCE GmbH</td>
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<td>10070051</td>
<td>CAVOK</td>
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<td>10070054</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
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<td>10070067</td>
<td>FOKKER TECHNIK B.V.</td>
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<td>10070084</td>
<td>STG AEROSPACE GERMANY GmbH</td>
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**List of EASA Supplemental Type Certificates**

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<th>STC No.</th>
<th>STC Holder</th>
<th>FAA STC</th>
<th>Description</th>
<th>Certification Basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tr>
<td>1007008</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA ST ST04078NY</td>
<td>Update of go-pro 2ku</td>
<td>1. Delta Engineering Master Data List 0165-10683-2498 Revision A-dated 18 April 2019 or later FAA approved.</td>
<td>STC ref.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/06/2019</td>
<td>Active</td>
<td>10070007</td>
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<td>1007009</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>ERJ 190-200 STD</td>
<td>MTV-27-() on Beech 1900</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST04078NY.</td>
<td>- According to MT-Propeller Master Data List No. E-3389 Rev. 0- To be installed i.a.w. Installation Instruction Doc. No. E-3387 Rev. 0- To be maintained i.a.w. MT- Propeller ICA Document No. E-3388 Rev. for later...</td>
<td>03/06/2019</td>
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<tr>
<td>10070091</td>
<td>365 AEROSPACE DDS LIMITED</td>
<td>FAA ST NO. ST04369NY</td>
<td>Change to introduce a modification to Zodiac Standard Elite Doublesets Rev. 2, Transfer of STC to new legal entity.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment:</td>
<td>STC ref.</td>
<td>According to MT-Propeller Master Data List No. MDL-0631 Revision A-365 Aerospace Instructions for Continued Airworthiness ICA-0631-01 – Change to introduce a modification to Zodiac Standard Elite/Doublesets (v, v) Configuration 1' Issue A-365 Aerospace Instructions for Continued Airworthiness ICA-0631-0...</td>
<td>24/10/2019</td>
<td>Active</td>
<td>10070009</td>
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<tr>
<td>10070096</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>T67M-MKI FIREFLY</td>
<td>Installation of MTV-12-( ) Propeller on Slingsby T67M Firefly Aircraft Models.</td>
<td>Change/Repair and any interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>STC ref.</td>
<td>Refer to 365 Aerospace Master Data List No. MDL-0631 for configuration standards that are a pre-requisite for this modification. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/06/2019</td>
<td>Active</td>
<td>10070010</td>
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<td>10070100</td>
<td>GDC TECHNICS, LLC.</td>
<td>FAA STC ST03109NY</td>
<td>Installation of Executive interior in an Airbus A330-200 Series This modification consists in the installation of an executive interrior which provides ninety eight (98) TaXi/Take-Off/Landing (TTL) passengers seats and eleven (11) TTL cabin attendant seats and includes a Panasonic Ax2 and Global Communica...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or alternative airworthiness requirements: Special Condition (s): D:4302-004 isolated Components D:4330-002 Glass/Screen of Displays / Monitors; D:4314-004 Shower Installation/Deviation(s): D:4301-004 Installation...</td>
<td>Master Data List, ref. 08873 Rev B dated 13 August 2014/Aircraft Flight Manual Supplement, ref. 579-TP-49 Revision A dated 11/November 2014/Installation Instruction Doc. for Continued Airworthiness, ref. 074-0876-2370 Rev cas A dated 22 August 2014/for later revisions of the above listed document(s) approved/accepted on...</td>
<td>The following modifications shall be installed concurrently: <strong>EASA STC 10070167</strong> (equivalent to FAA STC ST07284NY); <strong>Panasonic KXBand installation- EASA STC 10043698</strong> (equivalent to FAA STC ST03061CH); <strong>TFP tolerance- EASA STC 10032264</strong> (equivalent to FAA STC ST02317A) **Installation of Cabin interio...</td>
<td>11/02/2021</td>
<td>Active</td>
<td>10070101</td>
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<td>10070115</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>777-200, 777-300ER</td>
<td>Boeing 777-200 and 777-300ER aircrafts – tolerance to high electromagnetic emissions of TPEDs (Transmitting Personal Electronic Devices).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or alternative airworthiness requirements: Special Condition (s): D:4302-004 isolated Components D:4330-002 Glass/Screen of Displays / Monitors; D:4314-004 Shower Installation/Deviation(s): D:4301-004 Installation...</td>
<td>Master Data List, ref. 08873 Rev B dated 13 August 2014/Aircraft Flight Manual Supplement, ref. 579-TP-49 Revision A dated 11/November 2014/Installation Instruction Doc. for Continued Airworthiness, ref. 074-0876-2370 Rev cas A dated 22 August 2014/for later revisions of the above listed document(s) approved/accepted on...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/06/2019</td>
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<td>FAA STC</td>
<td>Rev. Level</td>
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<td>Associated Technical Documentation</td>
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<td>10070116</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC</td>
<td>SRO864DE</td>
<td>3806I</td>
<td>3606I</td>
<td>Installation of Concorde Battery Corporation 24 Ah VRRA Main Battery.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SRO864DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>1. Document No. S-5067, Master Drawing List, MD Helicopters/3606I/0000/0000 Battery Upgrade, Revision NC dated 1 June 2017, FAA approved 4 September 2017 or later</td>
<td>FAA approved revision 2. Instructions for Continued Airworthiness (CA), Document No. 5-05910-Revision A dated 31 May 2017, FAA accepted 31 Aug...</td>
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<td>10070122</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>FAA STC</td>
<td>ST03416NY</td>
<td>053</td>
<td>052</td>
<td>Installation of Concorde Battery Corporation 24 Ah VRRA Main Battery.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SRO864DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>1. Master Document, MDL-13218-0001, Issue 1 dated 3 June 2019—Instructions for Continued Airworthiness, 13218-ICA-0001, Issue 1 dated 6 December 2018—Change Bulletin, MOD-13218-0001, Issue 1 dated 6 December 2018for later revisions of the above listed document(s) approved/accepted under the EASA...</td>
<td>Approval limited to S-76 C helicopters equipped with two Turbomeca Arriel 2S2 engines with dual FADEC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC</td>
<td>ST03416NY</td>
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<td>Installation of a gogo 2Ku radome/antenna with demod option. Configuration 1: A319 ODA-2407-01 Configuration 2: A319-2150 ODA-2407-01 Configuration 3: Demod of 2Ku Configuration 3a: Demod of ATG Antenna Configuration 3b: Demod of Live TV A310 Configuration 3c: Demod of Panasonic D65 (DTV) System A310...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SRO864DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>5: Aircraft Flight Manual Supplement, ref: 06/04-06813-2407 Revision B dated 14 June 2022; For configurations 1, 2, 4, 5: Aircraft Flight Manual Supplement, ref: 06/04-06813-2407 Revision B dated 14 June 2022; For configurations 1, 2, 3, 3b, 3c, 3d, 3e, 3f, 4, 5: Instructions for Continued Airworthiness, ref: 07/04-06813-24...</td>
<td>As per MDL-1564-06813-2407 and ICA document 0764-06913-2407, incorporating Additional airworthiness approval is necessary for the configuration(s) explicitly covering the activation of the 2Ku antenna—Configuration 3b requires existing Live TV radome, fittings, antenna and adapter plate previously P...</td>
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<td>10070124</td>
<td>STAR WORK SKY S.A.S.</td>
<td>FAA STC</td>
<td>ST03416NY</td>
<td>053</td>
<td>052</td>
<td>Installation of a gogo 2Ku radome/antenna with demod option. Configuration 1: A319 ODA-2407-01 Configuration 2: A319-2150 ODA-2407-01 Configuration 3: Demod of 2Ku Configuration 3a: Demod of ATG Antenna Configuration 3b: Demod of Live TV A310 Configuration 3c: Demod of Panasonic D65 (DTV) System A310...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SRO864DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>5: Aircraft Flight Manual Supplement, ref: 06/04-06813-2407 Revision B dated 14 June 2022; For configurations 1, 2, 4, 5: Aircraft Flight Manual Supplement, ref: 06/04-06813-2407 Revision B dated 14 June 2022; For configurations 1, 2, 3, 3b, 3c, 3d, 3e, 3f, 4, 5: Instructions for Continued Airworthiness, ref: 07/04-06813-24...</td>
<td>As per MDL-1564-06813-2407 and ICA document 0764-06913-2407, incorporating Additional airworthiness approval is necessary for the configuration(s) explicitly covering the activation of the 2Ku antenna—Configuration 3b requires existing Live TV radome, fittings, antenna and adapter plate previously P...</td>
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<td>10070128</td>
<td>ANORA INTERNATIONAL LLC</td>
<td>FAA STC</td>
<td>ST02605LA</td>
<td>053</td>
<td>052</td>
<td>Installation of a gogo 2Ku radome/antenna with demod option. Configuration 1: A319 ODA-2407-01 Configuration 2: A319-2150 ODA-2407-01 Configuration 3: Demod of 2Ku Configuration 3a: Demod of ATG Antenna Configuration 3b: Demod of Live TV A310 Configuration 3c: Demod of Panasonic D65 (DTV) System A310...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SRO864DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>5: Aircraft Flight Manual Supplement, ref: 06/04-06813-2407 Revision B dated 14 June 2022; For configurations 1, 2, 4, 5: Aircraft Flight Manual Supplement, ref: 06/04-06813-2407 Revision B dated 14 June 2022; For configurations 1, 2, 3, 3b, 3c, 3d, 3e, 3f, 4, 5: Instructions for Continued Airworthiness, ref: 07/04-06813-24...</td>
<td>As per MDL-1564-06813-2407 and ICA document 0764-06913-2407, incorporating Additional airworthiness approval is necessary for the configuration(s) explicitly covering the activation of the 2Ku antenna—Configuration 3b requires existing Live TV radome, fittings, antenna and adapter plate previously P...</td>
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<td>STGC</td>
<td>Rev. Level</td>
<td>Foreign/NA STC ref.</td>
<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>STG AEROSPACE GERMANY GmbH</td>
<td>10070149</td>
<td>REV. 3</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>saf-Topfl SILLS.23 PHOTOLUMINESCENT AISLE PATH MARKING SYSTEM Revision 2: Adding of Associated Technical Documentation Revision 3: Transfer of STC holder from STG A erospace Limited to STG A erospace GmbH</td>
<td>The Certification Basis for the original product as amended by the following: a) Original STC 25 or Amendment 22.25.1.1.e, 25.26.a(6), 25.27.253.301, 25.303, 25.305; 25.307.25.58(1a)(c), 25.581, 25.601, 25.603, 25.609, 25.611, 25.613, 25.786(a); 25...</td>
<td>EASA STC holder to STG A erospace LTD AIRPLANE FLIGHT MANUAL SUPPLEMENT No. SG-1026- FMS-93 Revision IR (AirBib) STG A erospace LTD AIRPLANE FLIGHT MANUAL SUPPLEMENT No. SG-1026- FMS-93 Revision IR (AER) STG A erospace LTD AIRPLANE FLIGHT MANUAL SUPPLEMENT No. SG-1026- FMS-93 Revision IR (CRJ) STG A erospace LTD AIRPLANE FLIGHT MANUAL SUPPLEMENT No. SG-1026- FMS-93 Revision CRJ</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2021</td>
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<td>DASSAULT FALCON SERVICE</td>
<td>10070152</td>
<td>REV. 2</td>
<td>MYSTERE FALCON 900 (MF900); MYSTERE FALCON 50 (MF50); FALCON 900EX (F900EX); FALCON 2000EX, FALCON 7X</td>
<td>True North DLU satcom Activation Rev 1: extension applicability to the Falcon 900EX, MF50, MF900Rev 2: extension applicability to the Falcon 7X</td>
<td>The Certification Basis for the original product as amended by the following: a) Original STC 25 or Amendment 22.25.1.1.e, 25.26.a(6), 25.27.253.301, 25.303, 25.305; 25.307.25.58(1a)(c), 25.581, 25.601, 25.603, 25.609, 25.611, 25.613, 25.786(a); 25...</td>
<td>Master Data List DFS-DE-17-AML-P001 Issue C or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>The minor modification DFS-2061A (or later approved issue) on Falcon2000EX Post-mod (M1691, M1745 and M1504) and the modification DFS-9871Aon F900EX EASy Post-mod (M3058) and DFS-1177A (or later issue) on Falcon 7X are pre-required for this design change.</td>
<td>07/07/2020</td>
<td>Active</td>
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<td>AEROSERVIS, S.R.O.</td>
<td>10070153</td>
<td>REV. 1</td>
<td>L-410 UW-P6</td>
<td>STC AE187/A01470 release of MTOW to 6600kgs</td>
<td>The Certification Basis for the original product as amended by the following: a) Original STC 25 or Amendment 22.25.1.1.e, 25.26.a(6), 25.27.253.301, 25.303, 25.305; 25.307.25.58(1a)(c), 25.581, 25.601, 25.603, 25.609, 25.611, 25.613, 25.786(a); 25...</td>
<td>Flight Manual Supplement(x)(y)(z)(a)/STCAE/187-DLP1, Apr 10/19, MASTER DOCUMENT LIST, AE187/001, RO, Data 24.05.2019, Definition Document, AE187/01F, RC, Date 11.04.2019 or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/06/2019</td>
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<td>airplus engineering GmbH</td>
<td>10070164</td>
<td>REV. 1</td>
<td>DHC-6 SERIES 400</td>
<td>Installation of Honeywell Go Direct Router. This modification enables IFI functionality within the aircraft cabin. The equipment will be interfaced with the Aspro 200 system and AMT-1800G SatCom Antenna. See Description of Modification, doc. no. DE-2231-000001 Iss 1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Master Document List MD-2231-00000, Issue 3- Aircraft Flight Manual Supplement PO-2231-00000, Issue 0-1-Maintenance Manual Supplement MA-2231-00000, Issue 1 - Engineering Work OrderEW-2231-00000, Issue 2 or later revisions of the above list...</td>
<td>This modification is only applicable to aircraft with Honeywell Aspro200 System (Airplus Minor Change 2122) and Honeywell AMT-1800/Intermediate Gain Antenna (EASA STC 10065369) installed. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2021</td>
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<td>AEROLEDs, LLC</td>
<td>10070175</td>
<td>REV. 1</td>
<td>FAA STC SA02386AK</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>1. AeroLEDs Master Drawing List (MDL) document as listed onAML SA02386AK and later FAA approved revisions. 2. Instructions for Continued Airworthiness (ICA) document as listed onAML SA02386AK and later FAA approved revisions.</td>
<td>This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List SA02386AK dated 11 March 2015 or later approved revisions. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherprevi...</td>
<td>13/06/2019</td>
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<tr>
<td>ELDEC AEROSPACE CORPORATION</td>
<td>FAA STC ST004215E</td>
<td>SEE FAA APPROVED MODEL LIST: Installation of a Crane Aerospace &amp; Electronics Tire Sensor Assembly</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. Requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List ST024215E dated 14 January 2015 or later approved revisions Prior to installation of this change/repair it must be determined that the interoperability between this change/repair and any others...</td>
<td>13/06/2019</td>
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<td>JETTECH, L.L.C.</td>
<td>FAA STC ST01897WI</td>
<td>500, 550, 560, 5550: Installation of the RVSM capable Garmin G700 with GDU-1060 Touch Screen/Primary Flight Display (PFD).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC ST01897W. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>Prior to installation of this design change it must be determined that the interoperability between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>14/06/2019</td>
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<tr>
<td>JETTECH, L.L.C.</td>
<td>FAA STC ST01898WI</td>
<td>501, 551: Installation of the RVSM capable Garmin G600 with GDU-1060 Touch Screen/Primary Flight Display (PFD).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC ST01898W. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>Prior to installation of this design change it must be determined that the interoperability between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
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<td>AIRCRAFT RESTORATION CORP.</td>
<td>FAA SA00555DE</td>
<td>SEE FAA APPROVED AML SA00555DE: Replacement of the interior main landing gear door with an aerodynamic Fairing.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA00555DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>Prior to installation of this design change it must be determined that the interoperability between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>13/06/2019</td>
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<td>FLYING COLOURS CORP.</td>
<td>BD-700-1A10</td>
<td>Cabin Reconfiguration installation to be completed in accordance with Modification DataSummary L688000 Revision NC or later approved revision.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval. Installation to be completed in accordance with Modification Data Summary L688000 Revision NC. Refer to Instructions for Continued Airworthiness L688070 Revision NC or later revisions of the above listed document/s approved/accepted on behalf of EASA in accordance with the Technical Implementation Proc...</td>
<td>As per original Transport Canada STC C-L135A1—4010, this cabin reconfiguration is limited to BD-700-1A10 aircraft serial number 9189. Prior to installation of this change/repair it must be determined that the interoperability between this change/repair and any other previously installed change and/or...</td>
<td>14/06/2019</td>
<td>Active</td>
<td>10070201</td>
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**Notes:**
- The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.
- The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval...
- Prior to installation of this design change it must be determined that the interoperability between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product...

**References:**
- FAA STC: FAA Supplemental Type Certificates
- EASA: European Union Aviation Safety Agency
- CR070157: Certification Basis (CB) for the original product remains applicable to this certificate/ approval.
- CR070158: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070159: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070160: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070161: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070162: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070163: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070164: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070165: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070166: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070167: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070168: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070169: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070170: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070171: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070172: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070173: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070174: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070175: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070176: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
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- CR070187: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070188: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070189: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070190: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070191: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070192: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070193: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070194: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070195: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070196: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070197: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070198: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070199: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070200: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...
- CR070201: Certification Basis (CB) for the original product remains applicable to this certificate/ approval...

**Technical Information:**
- COLOURS CORP. incorporated 1974, with headquarters in Canada
- JETTECH, L.L.C., incorporated 1984, with headquarters in the United States
- AIRCRAFT RESTORATION CORP.
- FLYING COLOURS CORP.

**Links:**
- [EASA.europa.eu](https://easa.europa.eu)
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<tr>
<td>10070206</td>
<td>ALPINE AVIATION, INC.</td>
<td>FAA SA0095DE</td>
<td>1900D</td>
<td>Installation of a Cargo Interior.</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA0095DE.</td>
<td>1. Master Drawing List Document No. MDL-2015-01, Revision B, dated 15 October 2018, or later FAA approved revision. 2. Aircraft Flight Manual Supplement Document No. AFMS-2015, NoRevisions, dated 24 September 2018, or later FAA approved revision 3. Instructions for Continued Airworthiness: Document No.R.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/06/2019</td>
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<td>10070206</td>
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<td>10070212</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A320-232</td>
<td>Certification for commercial operation.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: Appendix G to CS-25 Amendment 21. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- The change is defined with Revision A of Master Data List Document MDL- EDD3-2500. The aircraft shall be operated in accordance with Revision A of obliqueplane Flight Manual Supplement AFMS- EDD3-2500 late revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/06/2019</td>
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<td>10070212</td>
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<td>10070213</td>
<td>SAFRAN SEATS</td>
<td>REV. 1</td>
<td>777-300ER &amp; 787-9</td>
<td>Upper Class Suite (UCS) Refresh Programme Rev. 1: Transfer from Safiran Seats GB Limited to Safiran Seats.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- SAFRAN Seats GB Instructions for Continued Airworthiness No. C1343CRevision 0- SAFRAN Seats GB Master Data List No C1343MLD Revision 0 for later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>1. The existing approved cabin layouts were approved by the FAA at original time of aircraft integration (refer to Master Data List No C1343MDL). Should these configurations be introduced in the EU, the baseline cabin configuration(s) will need to be approved by the Agency. The approval of this change:</td>
<td>09/07/2019</td>
<td>Active</td>
<td>10070213</td>
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<td>10070222</td>
<td>AIRBUS HELICOPTEROS, Inc</td>
<td>FAA SR11169DS</td>
<td>AS 350 B3</td>
<td>Installation of provisions to install Garmin G500H GDU 1066Display, L3ESI-500 Electronic Standby Indicator and TAU20 USB Charging Port.</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SR11169DS.</td>
<td>1. Airbus Helicopters Inc Master Drawing List (MDL) 355A02-1904 Rev. Dated 23 January 2019 or later ODA approved revision 2. Configuration: 01- Rotorcraft Flight Manual Supplement (RFMS) 350FMS-556 Rev. IR dated 31 January 2019 or later ODA approved revision 3. Instructions for Continued Airworthiness:</td>
<td>1. This STC is limited to Airbus Helicopters AS350 B3 Post Mod 07-4723 or Post Mod 07-4961 aircraft only. 2. Installation of Garmin G500H TXI Display System (Primary Flight Display (PFD), Multi-Function Display (MFD)) AML STC 10069864 is a prerequisite for installing this STC.3.</td>
<td>17/06/2019</td>
<td>Active</td>
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List of EASA Supplemental Type Certificates

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<tr>
<td>10070239</td>
<td>APTOZ EHF</td>
<td>REV. 1</td>
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<td>747-400F</td>
<td>Installation of GPS/ SBAS sensors/ antennae and installation of two (TSO-145c) CMC Eterline CMA-3024 GPS/ SBAS sensors, (TSO-C119) antennas, and associated wiring. The installed equipment underling will be inactive, activation is via separate STC. Reason for Rev. 1: Addition of 747-100, -200 and -300 models</td>
<td>The Certification Basis for the original product as amended by following additional or alternative airworthiness requirements: Special Condition CS-25 H-01 The following paragraph (s) at a later amendment: CS-25 amendment 4: CS25-29, CS 25.301, CS 25.303, CS 25.305, CS 25.307, CS 25.337, CS 25.341 CS...</td>
<td>- Master Data List document No: MDL341017, Rev. 2, (issued) Feb 2021 - Instructions for Continued Airworthiness documents No: EASA 19017, n&quot;m&quot; corresponds to the customised version related to the specific/applicable type, refer to respective table at the MDL or later revisions of the above listed docs...</td>
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<td>10070247</td>
<td>Falcon Crest Aviation Supply, Inc.</td>
<td>FAA STC</td>
<td>SRO0098RC</td>
<td>206L-4, 206L-3, 206L-1</td>
<td>Installation of a Concorde RG222 Valve Regulated Sealed Lead Acid Battery</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Instructions for Continued Airworthiness documents No: EASA 19017, n&quot;m&quot; corresponds to the customised version related to the specific/applicable type, refer to respective table at the MDL or later revisions of the above listed docs...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10070255</td>
<td>Aerotec &amp; Concept</td>
<td>FAA STC</td>
<td>SRO0098RC</td>
<td>A320-271N, A321-231-232/-233/-235/-251N/-252N</td>
<td>Installation of a Rain Repellent Dry Coat Application on the Outside Surface of SAINT-GOBAIN SULLY (GSS) Window</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Instructions for Continued Airworthiness documents No: EASA 19017, n&quot;m&quot; corresponds to the customised version related to the specific/applicable type, refer to respective table at the MDL or later revisions of the above listed docs...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10070280</td>
<td>Alpha Aviation, Inc.</td>
<td>FAA STC</td>
<td>SAO2211CH</td>
<td>Installation of a Three Point Torso Restraint System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Alpha Aviation Inc. Installation and Maintenance Manual as listed on AML, SAO2211CH, released 20 December 2016, or later FAA Approved revisions.</td>
<td>The Three Point Torso Restraint System installation is to be inspected/evaluated in accordance with Alpha Aviation Inc. Installation and Maintenance Manual as listed on AML, SAK2211CH, released 20 December 2016, or later FAA Approved revisions. This STC is valid for aircraft types and models approved...</td>
<td>21/06/2019</td>
<td>Active</td>
<td>10070280</td>
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<td>10070267</td>
<td>Standard Aero Business Aviationservices ES, L.L.C.</td>
<td>FAA STC</td>
<td>ST04316CH</td>
<td>EMB-145SL, EMB-145MP, EMB-145EP, EMB-145ER, EMB-135ER, EMB-135SLR, EMB-145</td>
<td>Installation of a Smartsky System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST04316CH. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval.</td>
<td>- Standard Aero Master Data List 103470, Revisions (C), dated 23 March 2019, or later FAA approved revision... - Airplane Flight Manual Supplement Document Number 1036750, Revisions (B), dated 20 March 2019, or later FAA approved revision... - Instructions for Continued Airworthiness documents No: EASA 19017, n&quot;m&quot; corresponds to the customised version related to the specific/applicable type, refer to respective table at the MDL or later revisions of the above listed docs...</td>
<td>Prior to installation of this change/design change it must be determined that the interrelationship between this change/design change and any other previously installed change/design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/06/2019</td>
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<td>10070267</td>
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**List of EASA Supplemental Type Certificates**

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<tr>
<th>STC Holder</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td><strong>STC Holder</strong></td>
<td><strong>Description</strong></td>
<td><strong>Certification basis</strong></td>
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<td><strong>Limitations/Conditions</strong></td>
<td><strong>Date of Issue</strong></td>
<td><strong>Status</strong></td>
<td><strong>System Record</strong></td>
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<tr>
<td>1007023</td>
<td>AS 332 L.1</td>
<td>Beckav Aeronica BKT3515 series. Transponder with Mode S. Elementary Surveillance and Automatic Dependent Surveillance – Broadcast (ADS-B). Out capabilities (ref. modification AHSF-70870).</td>
<td>Certification Programme (CP) ref. AHSF-70873 Revision B, dated 09/05/2019. CS-ACNS Compliance Plan ref. AHSF-70881 Initial issue, dated 09/05/2019. AS 332 L.1 Flight Manual Supplement (FMS) ref. HTnis 032 Revision dated 20/06/2019 Product Change Declaration ref. AHSF-70870 Revision A, dated 21/06/2019.</td>
<td>“ADS-B / MODE S OFF” function is outside of the scope of the CT and part of the EASA approved configuration. Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will not introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/06/2019</td>
<td>Active</td>
<td>1070727</td>
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<tr>
<td>1007024</td>
<td>REV. 3 FAA STC: ST01618WI-D</td>
<td>Installation of a Honeywell Enhanced Ground Proximity Warning System (EGPWS) and Activation of the Smart Runway/Smart Landing Features in accordance with Honeywell Master Drawing List 795-0251-002EASA validation of FAA STC ST01618WI-D Configuration 2.</td>
<td>Honeywell Master Drawing List 795-0251-002 Revision A dated 31/08/2019.</td>
<td>- EGPWS 95/0976-003-XXX-XXX MK V with software version 212.212 or later installed is a pre-requisite to this modification. Limitations: per Airplane Flight Manual Supplement 795-0251-811 Revision A dated 25 February 2019 RA for Master Drawing List 795-0251-002 be the required instr...</td>
<td>16/10/2019</td>
<td>Active</td>
<td>1070724</td>
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<td>1007027</td>
<td>REV. 1 737-900ER, 737-8, 737-9, 737-800, 737-700, 737-700, 737-700, 737-700</td>
<td>Major change to STC 10070279KA Band Satcom System Installation. Amendment of the STC 10070279 to include the system installation on the B737-7 with doubler and fasteners patterns compatible with...</td>
<td>- Certification Plan: JESL-M-01106CER-01-4r after further revision of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>None</td>
<td>02/07/2024</td>
<td>Active</td>
<td>1070726</td>
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<td>1007028</td>
<td>REV. 1 FAA STC: ST01920WI</td>
<td>Installation of software Batch 3.4 to the Honeywell Integrated Avionics Computer (IAIC) model C-810 on a Model BD-700-1A10 and BD-700-1A11</td>
<td>Master Document List (MDL) BAS0701521 Revision (F), dated 10 May 2019. FAA approved 10 May 2019, or later FAA approved/or later revision FAA approved/or later revision approved Aircraft Flight Manual Supplement 795-0251-811 Revision A dated 21 February 2019, FAA accepted 10 May 2019, or later FAA accepted revision for Continued Aeronauticalworthiness: BAS0706501-21A-1 Revision (HC), dated 21 February 2019, FAA accepted 10 May 2019, or later FAA accepted revision Airplane Flight Manual Supplement (FMS) ref. 09/06/2019.</td>
<td>Prior to installation of this design change it must be determined that the interaction between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. The aircraft must be equipped with Honeywell Int...</td>
<td>01/07/2019</td>
<td>Active</td>
<td>1070721</td>
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<td>1007029</td>
<td>REV. 1 737-900ER, 737-8, 737-9, 737-800, 737-700, 737-700, 737-700, 737-700</td>
<td>The Type of OSiD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>Master Document List (MDL) BAS0701521 Revision (F), dated 10 May 2019. FAA approved 10 May 2019, or later FAA approved/or later revision FAA approved/or later revision approved Aircraft Flight Manual Supplement 795-0251-811 Revision A dated 21 February 2019, FAA accepted 10 May 2019, or later FAA accepted revision Airplane Flight Manual Supplement 795-0251-811 Revision A dated 21 February 2019, FAA accepted 10 May 2019, or later FAA accepted revision for Continued Aeronauticalworthiness: BAS0706501-21A-1 Revision (HC), dated 21 February 2019, FAA accepted 10 May 2019, or later FAA accepted revision Airplane Flight Manual Supplement (FMS) ref. 09/06/2019.</td>
<td>Prior to installation of this design change/repair it must be determined that the interaction between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/07/2019</td>
<td>Active</td>
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<tr>
<td>10070281</td>
<td>REV. 1 FAA STC: ST01920WI</td>
<td>Installation of software Batch 3.4 to the Honeywell Integrated Avionics Computer (IAIC) model C-810 on a Model BD-700-1A10 and BD-700-1A11</td>
<td>Master Document List (MDL) BAS0701521 Revision (F), dated 10 May 2019. FAA approved 10 May 2019, or later FAA approved/or later revision FAA approved/or later revision approved Aircraft Flight Manual Supplement 795-0251-811 Revision A dated 21 February 2019, FAA accepted 10 May 2019, or later FAA accepted revision Airplane Flight Manual Supplement (FMS) ref. 09/06/2019.</td>
<td>Prior to installation of this design change/repair it must be determined that the interaction between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/07/2019</td>
<td>Active</td>
<td>1070721</td>
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<tr>
<td>1007029</td>
<td>REV. 1 737-900ER, 737-8, 737-9, 737-800, 737-700, 737-700, 737-700, 737-700</td>
<td>The Type of OSiD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>Master Document List (MDL) BAS0701521 Revision (F), dated 10 May 2019. FAA approved 10 May 2019, or later FAA approved/or later revision FAA approved/or later revision approved Aircraft Flight Manual Supplement 795-0251-811 Revision A dated 21 February 2019, FAA accepted 10 May 2019, or later FAA accepted revision Airplane Flight Manual Supplement (FMS) ref. 09/06/2019.</td>
<td>Prior to installation of this design change/repair it must be determined that the interaction between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/07/2019</td>
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<tr>
<td>10070281</td>
<td>REV. 1 FAA STC: ST01920WI</td>
<td>Installation of software Batch 3.4 to the Honeywell Integrated Avionics Computer (IAIC) model C-810 on a Model BD-700-1A10 and BD-700-1A11</td>
<td>Master Document List (MDL) BAS0701521 Revision (F), dated 10 May 2019. FAA approved 10 May 2019, or later FAA approved/or later revision FAA approved/or later revision approved Aircraft Flight Manual Supplement 795-0251-811 Revision A dated 21 February 2019, FAA accepted 10 May 2019, or later FAA accepted revision Airplane Flight Manual Supplement (FMS) ref. 09/06/2019.</td>
<td>Prior to installation of this design change/repair it must be determined that the interaction between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/07/2019</td>
<td>Active</td>
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<td>1007029</td>
<td>REV. 1 737-900ER, 737-8, 737-9, 737-800, 737-700, 737-700, 737-700, 737-700</td>
<td>The Type of OSiD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>Master Document List (MDL) BAS0701521 Revision (F), dated 10 May 2019. FAA approved 10 May 2019, or later FAA approved/or later revision FAA approved/or later revision approved Aircraft Flight Manual Supplement 795-0251-811 Revision A dated 21 February 2019, FAA accepted 10 May 2019, or later FAA accepted revision Airplane Flight Manual Supplement (FMS) ref. 09/06/2019.</td>
<td>Prior to installation of this design change/repair it must be determined that the interaction between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/07/2019</td>
<td>Active</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAAA STC ref.</td>
<td>Models</td>
<td>Certification basis</td>
<td>Description</td>
<td>Associated Technical Documentation</td>
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<td>10070298</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA-ST0164LA-D</td>
<td>GV-SP</td>
<td>Installation of a 14-Passenger cabin interior.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI F.1-10 (Installation of anon-rechargeable lithium battery). The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/approval.</td>
<td>FAA-GULFSTREAMSTDT-2019-0550</td>
<td>This STC is limited to serial number 5500 only. Prior to installation of the change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10070304</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>REV. 1</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Introduction of ADS-B Out Function Rev: 1; Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: C.25 at Amendment 21; C.25.01, C.25.601, C.25.869, C.25.899, C.25.1301, C.25.1302, C.25.1307, C.25.1309, C.25.132...</td>
<td>AES Limited Flight Manual Supplement No. 78 Issue 1</td>
<td>The following pre-requisites provisioning modifications are required: - lobe embodied prior to or concurrently with this modification (see applicable modification for issue status). Boeing 737NG Series Aircraft: AES-737-0270 Issue 1. Boeing 747 Series Aircraft: AES-747-0268 Issue 1. Boeing 787 Series Aircraft: AES-787-0269 Issue 1AES Limited Master Data List No. AES-0...</td>
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<tr>
<td>10070313</td>
<td>LUFTHANSA TECHINIK AG</td>
<td>REV. 1</td>
<td>A330-343</td>
<td>SWISS-Ultra High Conf. CAS Installation installation of a TSO approved UTAS seat track mounted Ultra High Comfort Cabin Attendant Seat (UHC-CAS) in SWISS Airbus A330-300/400 aircrafts, by replacing the installed high comfort CAS</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental Type Definition Document STD05-0707 IssuRev 01:00 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>None</td>
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<td>10070355</td>
<td>T.P.S. S.P.A.</td>
<td>REV. 1</td>
<td>A9W169</td>
<td>Installation of medical rack and retainers (ref: change TPS169-03-06) Rev:1</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Program (CP) ref. TPS169-0300-Cpi-03 Issue 2 dated/06/2019 Compliance Check List (CCL) ref. TPS169-0300-CCL-03 Issue 1 dated/03/04/2019</td>
<td>This STC requires as prerequisite the embodiment of at least one of the following passenger cabin layouts defined in the Leonardo Helicopters AW169 RFM Supplement 40 (&quot;Stretcher installation layout&quot;) forward (P/W) single stretcher with PW facing seats (P/N 6F2560F02211 or P/N 6F2560F02211); or MIDDLE...</td>
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<tr>
<td>10070370</td>
<td>JORMAC AEROSPACE, INC.</td>
<td>REV. 1</td>
<td>FAA ST04079NY</td>
<td>787-8/ 787-9/ 787-10/ 787-8.787-9/ 787-10</td>
<td>Installation of Jormac Aerospace Door Locks Revision 1; Transfer of STC holders from Delta Engineering to Jormac Aerospace: amendment validation STC accordingly</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC STO4079NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental Type Definition Document STD04079NY IssuRev 1</td>
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<td>10070402</td>
<td>LUFTHANSA TECHINIK AG</td>
<td>REV. 1</td>
<td>777-300ER</td>
<td>LATAM B777 Business Class Cabin Upgrade</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): D-129 &quot;Incorporation of Inertia Locking Device in slim seats The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to the above listed document(s) approved/accepted under the EASA system.</td>
<td>Supplemental Type Definition Document STD04079NY-17-12M Rev01:0009Revisions for Continued Airworthiness document A-00-19-51-ICARev02-01:00 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>STC Holder</td>
<td>FAA/STC</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>STANDARD AERO BUSINESS</td>
<td>FAA STD</td>
<td>Installation of an Electronic Flight Instrument System (EFIS).</td>
<td>The Certification Basis for the original product remains applicable to the certificate of approval, except where amended by additional or later amendments if indicated on FAA STC SR04390CH. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate of approval.</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/07/2019</td>
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<td>AVIATIONSERVIC ES, L.L.C.</td>
<td>SR04390C</td>
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<td>LUFTHANSA AVCON</td>
<td>10070442</td>
<td>Einbau Zusatzpumpe SF25A, Anbau von Räumen an den Flügelrippen. Änderung Kraftstoffzufuhr und Kraftstoffanzeige.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate of approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate of approval.</td>
<td></td>
<td>Limited to SN 4520.</td>
<td>10/07/2019</td>
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<td>AERO DYNAMIX, Inc.</td>
<td>FAA</td>
<td>Installation of Night Vision Goggle (NVG) Compatibile Lighting System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate of approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate of approval.</td>
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<td>10/07/2019</td>
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<tr>
<td>AEROSANTEC AG</td>
<td>REV. 1</td>
<td>Removal of Ka-band antenna and systems. The modification is to remove the system components and structures/provisions of the Ka-band antenna system, including the in- cabin distribution system, that have been previously installed per EASA STC 10063537.</td>
<td>The Type and OES Certification Basis (CB) for the original product remains applicable to this certificate of approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate of approval.</td>
<td></td>
<td>With this modification, EASA STC 10063537 and 10063559 are removed. Translated AIM-MR III are Instruements for Continued Airworthiness A0-16-23-ICA Issue 02 Rev.02 of later revisions of the above listed document(s) approved/ acceptable under the EASA system.</td>
<td>07/07/2023</td>
<td>Active</td>
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<tr>
<td>LUFTHANSA INDUSTRIES, Inc.</td>
<td>REV. 1</td>
<td>Installation of Fuselage Underside Rail Mounting Provision to carry external payload.</td>
<td>The Certification Basis for the original product remains applicable to the certificate of approval, except where amended by additional or later amendments if indicated on FAA STC SA1902WI. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate of approval.</td>
<td></td>
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<td>15/07/2019</td>
<td>Active</td>
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<td>DIEHL AVIATION LALPHIEIEN GmbH</td>
<td>LBA</td>
<td>Installation of Wardrobe PIN 104119 and PIN 104119S</td>
<td>The Certification Basis for the original product remains applicable to the certificate of approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate of approval.</td>
<td></td>
<td>None</td>
<td>22/07/2019</td>
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<td>TA0294</td>
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List of EASA Supplemental Type Certificates

EASA STC no.

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<th>DIHLE AVIATION LAULHEIM GmbH</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NA/STC ref.</th>
<th>Movie(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>LBA TA0205</td>
<td>DHC-8-315, DHC-8-312, DHC-8-314, DHC-8-307, DHC-8-311</td>
<td>Installation of Galley PN 106049C and PN 106049D</td>
<td>The Certification Basis for the original product remains applicable to this STC/FAA approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this STC/FAA approval.</td>
<td>None</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse interrelationship between this change/repair and any other previously installed change and/or repair.</td>
<td>22/07/2019</td>
<td>Active</td>
<td>1007479</td>
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<td>QUIT TECHNOLOGY AEROSPACE, Inc.</td>
<td>F/A Aircraft</td>
<td>STD4444AT</td>
<td>GULFSTREAM G280</td>
<td>The Certification Basis for the original product remains applicable to this STC/FAA approval.</td>
<td>1. Quiet Technology Aerospace document &quot;Bombardier Challenger 300/350 and Gulfstream G280内饰集成内栏杆更换MASTER DATA LIST&quot;, Report No. QTA 6301, Rev. A, dated 4 January 2019 or later</td>
<td>FAA approved revision.2. &quot;Bombardier Challenger 300/350 and Gulfstream G280内饰集成内栏杆更换MASTER DATA LIST&quot;, Report No. QTA 6301, Rev. A, dated 4 January 2019 or later</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse interrelationship between this change/repair and any other previously installed change and/or repair.</td>
<td>12/07/2019</td>
<td>Active</td>
<td>1007475</td>
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<td>AIRBUS DEFENCE AND SPACE S.A.</td>
<td>FAA STC</td>
<td>REV. 1</td>
<td>A320-243</td>
<td>A320 MR/TTF FAO OSD-FC/STC 10064192Revision 1 corrects the STC reference.</td>
<td>The Certification Basis for the original product remains applicable to this STC/FAA approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this STC/FAA approval.</td>
<td>None</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse interrelationship between this change/repair and any other previously installed change and/or repair.</td>
<td>16/07/2019</td>
<td>Active</td>
<td>1007478</td>
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<tr>
<td>PEREGRINE LLC</td>
<td>FAA STO8829E</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>B375-553 transponder</td>
<td>Installation of Backer Avionics BXT6553</td>
<td>The Certification Basis for the original product remains applicable to this STC/FAA approval.</td>
<td>1. Master Drawing List E- E17-2002 Rev. C FAA, Approved 29May 2019 or later FAA approved revision.2. Instructions for Continued Airworthiness E- FF-17-0008 Rev. IR dated 9May2019 or later FAA approved revision.3. Airplane Flight Manual Supplement E-FF-17-0009 Rev. IR dated 10 May2019 or later</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse interrelationship between this change/repair and any other previously installed change and/or repair.</td>
<td>15/07/2019</td>
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<td>ETHAD AIRWAYS ENGINEERING L.L.C.</td>
<td>FAA STO4892E</td>
<td>REV. 3</td>
<td>A320-232</td>
<td>Overwater &amp; Non-Overwater modification/Removal and exchange of some components like emergency equipment and/or cable ties based on the airworthiness directives and the FAA approved Revision.1: this revision is issued to add further configurations to EA2-19251035-DC01 and to...</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following FAA approved revision.1: this revision is issued to add further configurations to EA2-19251035-DC01 and to...</td>
<td>FAA approved revision.2. Instructions for Continued Airworthiness E- FF-17-0008 Rev. IR dated 9May2019 or later FAA approved revision.3. Airplane Flight Manual Supplement E-FF-17-0009 Rev. IR dated 10 May2019 or later</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse interrelationship between this change/repair and any other previously installed change and/or repair.</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>FAA STO8492E</td>
<td>REV. 3</td>
<td>737-700</td>
<td>Replacement of Satcom Antenna HGA-7000Replacement of the existing SATCOM High Gain Antenna HGA-7000 with the new HGA-7001 mounted on the top fuselage (above the ceiling ST4263-100)and removal of the existing Beam Steering Unit.</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following FAA approved revision.1: this revision is issued to add further configurations to EA2-19251035-DC01 and to...</td>
<td>FAA approved revision.2. Instructions for Continued Airworthiness E- FF-17-0008 Rev. IR dated 9May2019 or later FAA approved revision.3. Airplane Flight Manual Supplement E-FF-17-0009 Rev. IR dated 10 May2019 or later</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse interrelationship between this change/repair and any other previously installed change and/or repair.</td>
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### List of EASA Supplemental Type Certificates

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<th>STC ref.</th>
<th>Description</th>
<th>Certification Basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<td>10070554</td>
<td>LB AIRCRAFT ENGINEERING LLC</td>
<td>737-500</td>
<td>Replacement of the dual Honeywell or Collins enhanced Mode S transponders with a dual ACES X5-90 Automatic Dependent Surveillance-Broadcast (ADS-B) Out System installation of ADS-B Out Functionality.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated in FAA STC ST04448AT. The requirements for environmental protection and the associated changes to noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>1. LB Aircraft Engineering LLC, Master Data List LB-727-500-MDL-Revision E, dated 25 March 2019, or later FAA approved revisions.2. FAA Approved Flight Manual Supplement, LB Aircraft Engineering-document LB-727-500-APM, Rev IR, dated 10 May 2019, or later FAA approved revisions.3. LB Aircraft Engin...</td>
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<td>10070561</td>
<td>AGROCLUB ROMANIEI</td>
<td>Ils-28B2</td>
<td>Installation of provisions for smoke detectors and pyrotechnics on parking tips.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval, except where amended by:</td>
<td>Installation of provisions for smoke detectors and pyrotechnics on parking tips in accordance with Aeroclubs Romania, document MDU2.MDL-002 at Revision 2, dated 06 July, 2017, Aircraft Flight and Maintenance Supplement, Aeroclub Romania document AR-02.F03.1012, dated April 2018 or later revisions.</td>
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<td>10070566</td>
<td>SKANDIA, INC.</td>
<td>PC-12, PC-12i45, PC-12i47</td>
<td>Installation of Thermo/Acoustic Kits</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated in FAA STC SA4304CH.</td>
<td>Skandia Master Data List, Report No. Document Number 7903040MDL001.Revision E, dated 25 April 2019, or later FAA approved revisions.</td>
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<td>10070588</td>
<td>AIR ALLIANCE GmbH</td>
<td>35, 35B, 31, 31A, 35, 36A</td>
<td>GTX 3000 Dual Transponder installation including ADS-B out</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval, except where amended by:</td>
<td>According to Air Alliance Master Document List No. 2017_006_J35_55_Form...</td>
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<td>10070589</td>
<td>AIR ALLIANCE GmbH</td>
<td>REV. 1</td>
<td>Installation of ISAS GPS Antenna</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval, except where amended by:</td>
<td>According to Air Alliance Master Document List No. 2019-006-MDL, Rev 03To be installed i.a.e. Air Alliance Installation Instruction Doc. No. 2019-006-IM, Initial Issue To be maintained i.a.e. Air Alliance ICA Document No. 2019-006-ICA, Initial issue later revisions of the above listed document(s) apply.</td>
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<td>ST ENGINEERING AEROSPACE LTD.</td>
<td>10070590</td>
<td>Passenger cabin reconfiguration for STAS Model 268 series/Economy Class seats</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.833(a)(h)(c), CS25.1529 at Amendment 20. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Aircraft Compliance Summary Report 366/ GAID053 Rev B dated 23. July 2019</td>
<td>Master Document List 366_AED_001 Rev C</td>
<td>MDL Bassai Ara, dated 22. July 2019</td>
<td>Instructions for Continued Airworthiness ICA-0374-001 Rev C, dated 09. July 2019</td>
<td>24/07/2019</td>
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<td>NORTHERN AVIONICS S.R.L.</td>
<td>10070592</td>
<td>Installation of Automatic Dependent Surveillance - Broadcast Out (ADS-B Out) system on Piaggio P.180 Avanti II</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-ACNS 1 (certification of an ADS-B Out Extendedsquitter &amp; ILS installation (Major Change) and its compliance to CS-ACNS). The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Project specification in accordance with Master Document List, MDL015Rev. 2, in accordance with installation instructions, II-NAV061Rev.</td>
<td>0K Maintenance in accordance with instructions for continued airworthiness ICA-NAV061 Rev.</td>
<td>Instruction for Continued Airworthiness B777-300 Issue A dated 07 Mar 1918CA076</td>
<td>24/07/2019</td>
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<td>AEROSCIENCE GmbH</td>
<td>10070596</td>
<td>Winglets for a DG-300 ELAN</td>
<td>The Certification Basis for the original product remains applicable to the certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>- Zeichnung Winglet-Nr.: D-300/10 vom 02.06.2016 - Compliance Checklist vom 15.03.2019 - Ergänzungen zum Flughandbuch vom 03.09.2019 - 1919CA076</td>
<td>22.July.2019</td>
<td>Master Data List Issue C</td>
<td>25/07/2019</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>10070605</td>
<td>A340-313</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>SOCITEV AIR FRANCE S.A. Flight Crew Operations Manual Supplement CM015SOCITEA3401_002015SOCITE AIR FRANCE S.A. Flight Crew Operating Manual Supplement CM015SOCITEA3401_002015socitefrance Seats</td>
<td>Prior to installation of this change/repair it must be determined if there is an interrelationship between this change/repair and any other previously installed change and/or repair that would introduce no adverse effect upon the airworthiness of the product.</td>
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<td>SHOCK TRAUMA AIR RESCUE SERVICE</td>
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<td>AW139</td>
<td>Installation of Technisonic TDFM-9100 Multiband P25 Airborne Transceiver System</td>
<td>The Certification Basis for the original product remains applicable to this certificate.</td>
<td>Modification Data Summary</td>
<td>No. 25 April 2018</td>
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<td>The Certification Basis for the following additional or alternative airworthiness requirements:</td>
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<td>AW139</td>
<td>Installation of Technisonic TDFM-9100 Multiband P25 Airborne Transceiver System</td>
<td>The Certification Basis for the original product remains applicable to this certificate.</td>
<td>Modification Data Summary</td>
<td>No. 25 April 2018</td>
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<td>Installation of Technisonic TDFM-9100 Multiband P25 Airborne Transceiver System</td>
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**An agency of the European Union**
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<td>ROW 44, Inc.</td>
<td>FAA ST03044NY</td>
<td>REV. 1</td>
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<td>737-700-800-900ER 737-700/ 737-800</td>
<td>Installation and activation of a new Row 44 Wireless Broadband Antenna System Rev. 1 adds the model 737-900ER.</td>
<td>Certification Basis for the original STC product remains applicable, with additional or later amendments if indicated on FAA STC ST03044NY. The requirements for environmental protection and the associated certified noise and emissions levels of the original STC product remain applicable to this certificate approval.</td>
<td>FAA Certification and/or Compliance Checklist (CDL) Rev. 1 adds the model 737-900ER.</td>
<td>09/04/2021</td>
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<td>ANCRA INTERNATIONAL LLC</td>
<td>FAA STC ST001SLA</td>
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<td>757-200</td>
<td>Installation of main deck cargo loading system</td>
<td>Certification Basis for the original STC product remains applicable, with additional or later amendments if indicated on FAA STC ST001SLA. The requirements for environmental protection and the associated certified noise and emissions levels of the original STC product remain applicable to this certificate approval.</td>
<td>FAA Certification and/or Compliance Checklist (CDL) Rev. 1 adds the model 757-200.</td>
<td>02/08/2019</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>EC 155 B, EC 155 B1</td>
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<td>767-300</td>
<td>LATAM Business Class Cabin Upgrade</td>
<td>Certification Basis for the original STC product remains applicable, with additional or later amendments if indicated on FAA STC ST001SLA. The requirements for environmental protection and the associated certified noise and emissions levels of the original STC product remain applicable to this certificate approval.</td>
<td>FAA Certification and/or Compliance Checklist (CDL) Rev. 1 adds the model 767-300.</td>
<td>31/07/2019</td>
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<td>SPAES GmbH &amp; Co. KG</td>
<td>EC 155 B, EC 155 B1</td>
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<td>155-208-25-01</td>
<td>Installation of Medical Equipment Ratainer PIN SDQ-155-208-25-01</td>
<td>Certification Basis for the original STC product remains applicable, with additional or later amendments if indicated on FAA STC ST001SLA. The requirements for environmental protection and the associated certified noise and emissions levels of the original STC product remain applicable to this certificate approval.</td>
<td>FAA Certification and/or Compliance Checklist (CDL) Rev. 1 adds the model 155-208-25-01.</td>
<td>31/07/2019</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST04077NY</td>
<td>REV. 1</td>
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<td>A330-941 A330-341, -342, -343 A330-321, -322, -323 A330-301, -302, -303 A330-223, -223F, -243, -245 A330-201, -202, -203 A330</td>
<td>Installation of a go2ku Cabin Connectivity System (Rev. 1 approves alternate location for the Ku radio frequency unit (KRFU)).</td>
<td>Certification Basis for the original STC product remains applicable, with additional or later amendments if indicated on FAA STC ST04077NY. The requirements for environmental protection and the associated certified noise and emissions levels of the original STC product remain applicable to this certificate approval.</td>
<td>Delta Engineering Master Data List (MDL) Revision A dated 28 December 2020.</td>
<td>28/02/2024</td>
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<td>10070657</td>
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<td>10070667</td>
<td>PEREGRINE LLC/PA PEREGRINE</td>
<td>FAA STC SR00925DE</td>
<td>SEE FAA APPROVED MODEL LIST</td>
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<td>10070669</td>
<td>ELDEC LLC</td>
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<td>10070670</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>FAA STC ST03311CH</td>
<td>BD-700-1A10, BD-700-1A11</td>
<td>Wireless Fidelity (Wi-Fi) Activation of a Previously Installed (IEEE802.11 b, g Only) Router and Transmitting Portable Electronic Device (T-PED) Aircraft Tolerance Testing</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/voideffect upon the airworthiness of the product. This STC is valid for aircraft types and models only</td>
<td>06/08/2019</td>
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<td>10070673</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>TCCA STC SH18-56 ISSUE 1</td>
<td>AS 350 B3, EC 130 T2</td>
<td>Mid-Continent M015-322 attimeter to replaceAerosoniccompassimeter PN H15025-0175</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/voideffect upon the airworthiness of the product. This STC is valid for aircraft types and models only</td>
<td>06/08/2019</td>
<td>Active</td>
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The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendment(s) if indicated on FAA STC SR00925DE. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendment(s) if indicated on FAA STC ST0432NY. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendment(s) if indicated on FAA STC ST03311CH. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendment(s) if indicated on FAA STC SH18-56 ISSUE 1. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.

Data Engineering Approved Model List 0198-10082-2419, Revision A0, dated 23 December 2021; and/or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.

See Specific Limitations § 2.2. of FAA Approved Model List. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/voice effect upon the airworthiness of the product.

See Specific Limitations § 2.2. of FAA Approved Model List. Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce noise/voice effect upon the airworthiness of the product.
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<th>STC Holder</th>
<th>FAA STC</th>
<th>Rev.</th>
<th>Foreign/NAA</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>S-TEC CORPORATION</td>
<td>172M, 172N, 172P, 172Q</td>
<td>FAA STC SAB925SW-D</td>
<td>Installation of S-TEC System 56/55X Two Axis Automatic Flight Guidance System, Model ST-663</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>- FAA STC 10070167</td>
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<td>BOMBARDIER Inc.</td>
<td>SEBOLD LUFTFAHRTTECH NISCHER BETRIEB(LTB)</td>
<td>REV. 2</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Airbrake spring element/Optional installation of a spring element in the airbrake push rod system for constant locking force and possibility of visual inspection of the locking force. Rev. 1 makes several drawings changes or corrections, to prevent mistakes due to missing information in manufacturing. Al...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Technical drawings: 1801B10E001.4-1801B10E002.3-1801B10E002.4-1801B10E003-1801B10E004-1801B10E005-1801B10E006</td>
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<td>BOMBARDIER inc.</td>
<td>TCXA STC SA19-10</td>
<td>BD-100-1A10, CL-660-2B16</td>
<td>Installation, Galway with Compartment for Commercial Appliances on Equipment EASA Validation of Transport Canada (TCXA) STC SA19-10 at issue 1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Modification must be in accordance with the following approved documentation a) For aircraft BD-100-1A10: B/E Aerospace Ltd Modification Data Summary L711001 Rev. NC, 27 June 2019, and b) For aircraft CL-660-2B16: B/E Aerospace Ltd Modification Data Summary L711002 Rev. NC, 27 June 2019.</td>
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<td>AVCON INDUSTRIES, Inc.</td>
<td>FAA STC SA01924WH</td>
<td>REV. 1</td>
<td>B220C B200 A200 B205 C 200, 200C A200, B200, B205C 200</td>
<td>Installation of an External Pod and Retractable EO/IR Sensor LiftMechanismRev. 1: adds a new configuration to the B220B/B205C models.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>1. Master Data List, Document No. 04607101, Rev. 5, dated 3 June 2019.</td>
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<tr>
<td>SIIL DEUTSCHLAND GMBH</td>
<td>EC135 T2</td>
<td>V450E-4K-Mosaic System Real-time airborne monitoring system consisting of external sensor pod installation to the multifunction deck, and installation of the processing unit in the cargo area.</td>
<td>The Certification Basis for the original product as amended The requirements for environmental protection and the associated certificated noise and/or emissions levels of the...</td>
<td>Compliance Program CP-239-00014-000, Issue 2</td>
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<td>21/02/2019</td>
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<td>AVCON INDUSTRIES, Inc.</td>
<td>B300C FAA STC SD1939W</td>
<td>Installation of a Retractable EC1/IR Sensor Mechanism.</td>
<td>The Certification Basis for the original product remains applicable with the associated Noise and/ or emissions levels of the...</td>
<td>1. Buffalo National Corporation Master List, No. 0401000, Rev. 1, dated 21 November 2018.</td>
<td>Prior to, or concurrent, installation of a new external sensor modification and any other previously installed change and/or repair will introduce no adverse effect upon the EASA system.</td>
<td>16/06/2019</td>
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<td>10070736</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>SB 340B</td>
<td>MODE S &amp; ADS-B UPGRADE The change consists of replacing the existing ADSB system with a new system capable of MOCES ELTs and ADS-B and OUTlink function. According to Minimum Operational Performance Standards (ADS-B Version 2 as per DO-268B including Conclusion 1 MOPS).</td>
<td>The Certification Basis for the original product as amended The requirements for environmental protection and the associated certificated noise and/or emissions levels of the...</td>
<td>1. Approved Aircraft Flight Manual Supplement, Document Ref. 2103002-AMA, Issue 1, dated 29.07.19. Continued Airworthiness, Document Ref. 2103002-CA-A, Issue 1, dated 28.07.19.</td>
<td>The requirements for environmental protection and the associated certificated noise and/or emissions levels of the...</td>
<td>16/06/2019</td>
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<td>MEDITERRANEAN AVIATION CORPORATION Ltd (MEDAVIA)</td>
<td>737-700</td>
<td>Maximum Take Off Mass Change in accordance with EASA record 21336.</td>
<td>The Certification Basis (CB) for the original product remains applicable with the associated Noise and/ or emissions levels of the...</td>
<td>Airplane Flight Manual Supplement MODEOE02811_AMS1 rev 2or later revisions of the...</td>
<td>The requirements for environmental protection and the associated certificated noise and/or emissions levels of the...</td>
<td>19/06/2019</td>
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<td>S-TEC CORPORATION</td>
<td>F172N FAA STC SA9134AC-D</td>
<td>Installation of S-TEC Flight Guidance System, Model ST-412</td>
<td>The Certification Basis for the original product as amended...</td>
<td>FAA/FAAS Approved Supplemental Material Record No. 217001145 Rev. 1, dated 24 October 2018.</td>
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<td>S-TEC CORPORATION</td>
<td>T303 FAA STC SA9754DS</td>
<td>Installation of S-TEC 0126 Digital Flight Control System (DFCS).</td>
<td>The Certification Basis for the original product remains applicable with the associated Noise and/or...</td>
<td>Master Data List No. ST-011-0001-000, Revision 2, dated 28 February 2001.</td>
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<td>S-TEC CORPORATION</td>
<td>10070757</td>
<td>Installation of a S-TEC 01326 Digital Flight Control System (DFCS). The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC 10070753D. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List No. ST-970-MDL-0001, Revision A, dated 3 April 2019, or later FAA approved revision. - Airplane Flight Manual Supplement, ST-970-AMS-0001, Revision A, dated 3 April 2019, or later FAA approved revision. - Instructions for Continued Airworthiness, ST-970-CAA-0001, Revision A, dated 6 March 2017.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/08/2019</td>
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<td>GROB AIRCRAFT SE</td>
<td>10070774</td>
<td>Approval of GROB Aircraft SE modification ref OCN C007-01 to install Dual Garmin GS Instruments. CS 23 Amendment 4: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Installation in accordance with GROB Aircraft SE modification ref OCN C007-01 to install Dual Garmin GS Instruments and Garmin STC 10060846. - Operation in accordance with Garmin approved Flight Manual Supplementation defined in EASA STC 10060846 and GROB Aircraft Flight Manual Supplement A1 Doc No G-C.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/08/2019</td>
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<tr>
<td>SIKORSKY AIRCRAFT CORPORATION</td>
<td>10070776</td>
<td>Installation of Center Console Four Inch Extension. The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SR02421NY-D. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Keystone Helicopter Installation Drawing List IDL01280, Rev. A, dated 26 June 2007, or later FAA approved revision. - Keystone Helicopter Instructions for Continued Airworthiness, Center/Console Extension Installation in Sikorsky S-92A Aircraft, Report No IC01260, Rev. A, dated 3 May 2007, or later.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/08/2019</td>
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<td>SPAES GmbH &amp; Co. KG</td>
<td>10070777</td>
<td>Installation of Nightscanner in tower forward fairing of MBB-BK117 Helicopters as mission equipment for searching and landing. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 29 ASDR A4 CS29.1316, CS 291317. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>EASA approved documents - Flight Manual Supplement, FMS-SPAES-0207, Revision 00 dated 10 July 2019. - Airworthiness Limitations Section, Section 8 of Instruction for Continued Airworthiness, ICA-117-207-33-01. - Airworthiness Limitations Section, Section 8 of Instruction for Continued Airworthiness, ICA-117-207-33-01, Appendix 01 dated 15 August 2019. - EASA accepted documents - Master Document List, MDL-117-207-3.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/08/2019</td>
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<td>R&amp;D PRODUCTS, Inc.</td>
<td>10070785</td>
<td>Installation of a Canadian Mancom (CMC) CMA-2012C in placed a RACAL 1900C Doppler The Certification Basis (CS) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Keystone Installation Drawing List 1DL01769 Rev. C, dated 21 April 2008 or later FAA approved revision. 2. Keystone Helicopter Instructions for Continued Airworthiness, CMICOMA-2012C Doppler Sensor Installed in a Sikorsky S-76C Aircraft, Report No. IC01769 Rev. C, dated 15 March 2008 or later FAA approved.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/08/2019</td>
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### List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
<th>STC No.</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tr>
<td>ALAMO ENGINEERING GmbH</td>
<td>CL-600-2B19 (RU SERIES 100)</td>
<td>REV. 1</td>
<td>-</td>
<td>-</td>
<td>Satcom Upgrade to SBB Aviator 700E including GSM and WLAN via SDR-G. The project consists in upgrading existing Immarsat AERO H+ SATCOM system to SBB “Aviator 700E” along with the installation of the SDR-Gigaloway (Satcom Direct) including Wi-Fi and 3G/4G functionality. Rev. 1: transfers the STC to...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Document List MDL.102301/1, Issue A, dated 21 August 2019 - Airplane Flight Manual Supplement AMIS.102312/1/001, Issue A, dated 22 July 2019 - Aircraft Maintenance Manual Supplement AMIS.102312/1/001, Issue A, dated 9 July 2019</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/12/2023</td>
<td>Active</td>
<td>10070706</td>
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<td>HELICOPTER FILMWORKS INC.</td>
<td>TC5A SH1-6</td>
<td>REV. 1</td>
<td>-</td>
<td>-</td>
<td>Installation of single and double mounts external cameras Rev. 1: correction of certificate.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Eligibility List CE-HFW-001, Rev 04 of April 27, 2017. Installation Instructions: a. C1-HFW-001-1, Rev. 3 dated March 26, 2017 (Single mount); b. C3-HFW-001-2, Rev 3 dated March 26, 2017 (Dual mount); 3. Instructions for Continued Airworthiness: a. CC-HFW-001-1, Rev. 3 dated March 26, 2017 (Single...</td>
<td>Day &amp; Night VFR only No CAT A operations permitted (for AS 355). Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the...</td>
<td>29/11/2019</td>
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<td>FREBALLONSPO RT VEREIN MUENSTERUND Muensterland e.V.</td>
<td>NL-STU/1000</td>
<td>-</td>
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<td>-</td>
<td>Leichtbaukorb Völker K-01</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-31GB initial issue. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Zulassungsprogramm EASA Projekt 0010002195 Völker K-01, dated 22 Aug, 2019, initial issue; - Flughandbuch Anhang Völker K-01, dated 22 Aug, 2019, initial issue; 5. later revisions of the above listed document(s) approved/accredited...</td>
<td>Other: Leichtbaukorb Völker K-01 kann nur mit Hülse Ballonbau Wörner NL-STU/1000 SN 1068 betrieben. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed...</td>
<td>23/06/2019</td>
<td>Active</td>
<td>10070707</td>
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<tr>
<td>DART AEROSPACE Ltd.</td>
<td>EC 130 T2 EC 130 B4</td>
<td>D130-1048 Landing Gear Kits</td>
<td>-</td>
<td>-</td>
<td>D130-1048 Landing Gear Kits</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Instructions: W1-D130-1048 Rev B, dated February 13, 2019 19/14/2019 (Single Mount); 3. Instructions for Continued Airworthiness: ICA-D130-1048 Revision 0, dated/December 20, 2019 Master Document ListMDL-D130-1048 Rev A, dated February 14, 2019(rev. of March 20, 2019) 9 of the above listed document(s) approved/accredited...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/06/2019</td>
<td>Active</td>
<td>10070709</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>A320-271N</td>
<td>REV. 1</td>
<td>-</td>
<td>-</td>
<td>Cabin conversion 188 to 180 passengerSeats at door 4L in A320 environment</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Equivalent Safety Finding(s): ESP D-2340-001 &quot;Dynamic seat qualification – Vertical test&quot;. The requirements for environmental protection and the associated certified noise a...</td>
<td>Compliance Check List CCL25-19/194M Revision 2 dated 29 May 2020, Compliance Check List CCL 25-19/10M Issue 7 dated 20 May 2019, Supplemental Type Design Document STD025-19/10M Issue 02 Revision 0 dated 25 May 2020, Supplemental Type Design Document STD025-19/10M Issue 01 Rev 00 dated 21 Aug 2019...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/05/2020</td>
<td>Active</td>
<td>10070801</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>Status</td>
<td>System Record</td>
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<td>FOKKER SERVICES B.V.</td>
<td>AGS-B on the Boeing 767: This change introduces AGS-B out capability on the 767. The ACFTransponders will be changed and they will interface with the GNSS position sensor.</td>
<td>Modification sheet EMA/17002/DR400/GNS430 Revision 3 dated 03.06.2019/Aeroplane FlightManual Supplement Edition 1 Revision 0 dated 27.02.2019 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/03/2019</td>
<td>Active</td>
<td>1007062</td>
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<tr>
<td>AEROMECHANIC</td>
<td>Installation of a Ki-204 and approval of GNS 430 for Night VFR Navigation</td>
<td>- Supplement to AFM SA19-025-AFMS Revision 2- Supplement to Aircraft Maintenance Manual SA19-025-AMMS Revision 1- Master Document List SA19-025-MDL-Revision 4- MMEL supplement SA19-025-MMELs rev for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/08/2019</td>
<td>Active</td>
<td>1007061</td>
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<td>SIRIUM AEROTECH S.L.</td>
<td>ADS-B System Upgrade for B737 Revisions 1: added reference toomitor change SA20-32M for MMEL-S</td>
<td>- Master Data List ref - MDL- S-23-4203 Issue A dated 20 August 2019- Instruction for Continued Airworthiness ref : ICAO-F3042-00-003 (Issue 3) dated 20 August 2019- Instruction for Continued Airworthiness ref : ICAO-F3065-23-006 Issue dated 13 August 2019 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The following modifications are prerequisites to this modification: - EASA STC 10032391 rev 2 - P710 VIP Cabin Completion- EASA STC 1005582 rev 4- Egide Safe Protection System (SPS) modification. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/09/2020</td>
<td>Active</td>
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<td>SABENA TECHNICS BOD</td>
<td>A320-223</td>
<td>- Master Data List ref - MDL- S-23-4203 Issue A, dated 20 August 2019- Instruction for Continued Airworthiness, ref : ICAO-F3042-00-003 (Issue 3) dated 20 August 2019- Instruction for Continued Airworthiness, ref : ICAO-F3065-23-006 Issue dated 13 August 2019 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The following modifications are prerequisites to this modification: - EASA STC 10032391 rev 2 - P710 VIP Cabin Completion- EASA STC 1005582 rev 4- Egide Safe Protection System (SPS) modification. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/08/2019</td>
<td>Active</td>
<td>1007061</td>
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<td>DART AEROSPACE Ltd</td>
<td>Dart D505-1057 Skid tube Installations</td>
<td>- Installation Instruction IMA- D505-1057 Rev. A dated 31 May 2019- Instruction for Continued Airworthiness ICA-D505-1057 Rev. 0 dated 28 April 2019- Master Document List MDL- D505-1057 Rev. A dated 03 June 2019 or later revisions of the above listed document(s) approved/accepted under EASA in acco...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/09/2019</td>
<td>Active</td>
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List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>Installation of MTV-37-1-E-C-F-R(G)CFR () on Cessna 441</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>According to MT-Propeller Master Documentation List No. E-3419 Rev. 0. To be operated in accordance with MT-Propeller/AFM-S Document No. E-3419 Rev. 0 To be installed i.a.w. Installation Instruction Doc. No. E-3419 Rev. 0. To be maintained i.a.w. MT-Propeller ICA Document No. E-3420 Rev. for later...</td>
<td>According limitation section of AFM-S Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/02/2019 Active</td>
<td>10070601</td>
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<tr>
<td>APUS - AERONAUTICAL ENGINEERING Gmb</td>
<td>F182P, F182Q 182S, 182T, T182T 182G, 182T, 182N Skydive Operation on C182 with 5 parachutists. The change is based on existing modifications for operation with 4 parachutists.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with APUS List of Design Definition Documents, No P70-A18131 at issue 00, dated 29 May 2019. Operation in accordance with APUS Aircraft Flight Manual Supplement, Nu41-900182 at revision 01, dated 03 May 2019 or later revisions of the above listed document(s) approved/accepted</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/09/2019 Active</td>
<td>10070603</td>
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<tr>
<td>FOKKER TECHNIEK B.V.</td>
<td>ATR 72-212 A Introduction of Immarsat and Iridium SATCOM system incl. WLAN system on ATR 72-212A ’600’ version/Revision 1: Transfer of STC from Fokker Services B.V. (EASA.21J.059) to Fokker Techniek B.V. (EASA.21J.667)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Compliance Record Report ref. CRR-0255 issue 1 dated 04 Sept. 2019 AFM Supplement ref. 19-D-014 issue 1 dated 12 July 2019 (ATR72TT/AFM-S-001) dated 11 July 2019 ICA Supplement ref. SE-790 issue 2 dated 18 July 2019 or later revisions of the above listed document(s) approved/accepted</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/12/2019 Active</td>
<td>10070611</td>
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<tr>
<td>HONEYWELL INTERNATIONAL Inc</td>
<td>FAA STC ST0704LA SEE FAA AML Installation of the Honeywell Fuselage Anti-Collision Light (FACL) System. Rev. 1 is a change of the certificate holder’s address and an addition of Boeing 757, 767 and 777 series models.</td>
<td>The Certification Basis for the original product remains applicable to the certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Honeywell Master Drawing List identified on the AML. The Instructions for Continued Airworthiness (ICA) identified on the AML.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/01/2021 Active</td>
<td>10070614</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC STC ref.</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
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<tr>
<td>10070938</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>500, 550, 560, 5550</td>
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<td>Line-1 upgrade for CPDLC and LPV. The purpose of this STC is to upgrade of an existing Universal FMISystem for the following options: 1. CPDLC communication installation, 2. CPDLC communication installation greater than 2.00km installation. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 2 - CS 25.31, CS 25.79, CS 25.301, CS 25.303, CS 25.305, CS 25.307, CS 25.314, CS 25.361, CS 25.581, CS 25.66.</td>
<td>Instructions for Continued Airworthiness for CPDLC, Doc: 211121-1CA-A, Issue 1. Instructions for Continued Airworthiness for CPDLC, Doc: 211121-1CA-A, Issue 1.</td>
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<tr>
<td>10070943</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>FAA STC SA01896WI</td>
<td>REV. 2</td>
<td></td>
<td>510</td>
<td>Approval of installation of Garmin G1000 NXi Integrated Avionics System Hardware and Software Updates, including Optional Flight Stream 510. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Based on Regulation 748/2012 Annex I (Part 21), and the EASA policy for significant changes, the certification basis for this change is as follows: The certification basis...</td>
<td>Installation in accordance with Garmin Master Document List (MDL) Number 005-01236-00 revision 1.0 dated 4 February 2019, and prior to installation of this change/repair it must be determined that the technical interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10070952</td>
<td>AYESA AIR CONTROL INGENIERIAERO NAUTICA SL</td>
<td>FALCON 10</td>
<td></td>
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<td>REPLACEMENT OF CVR IN FALCON 10. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level(s) and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AA19B1827YCP01MD0101 MASTER CERTIFICATION DOCUMENT LIST ISSUE 1AC71827YCP10MM0101 1 MAINTENANCE MANUAL SUPPLEMENT ISSUE for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<tr>
<td>10070963</td>
<td>SABENA TECHNICS BGC</td>
<td>FALCON 10</td>
<td></td>
<td></td>
<td>737-800</td>
<td>REPLACEMENT OF CVR IN FALCON 10. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level(s) and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Stream 510. and Software Updates, including Optional NXi Integrated Avionics System Hardware Installation in accordance with Garmin Master Document List (MDL) Number 005-01236-00 revision 1.0 dated 4 February 2019, and prior to installation of this change/repair it must be determined that the technical interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10070965</td>
<td>BOMBARDIER Inc.</td>
<td>TCCA Q-LSA19-0315D</td>
<td></td>
<td></td>
<td>BD-700-1A10</td>
<td>Cabin Refurbishment and CMS Upgrade The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level(s) and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List, ref: 40186-02-A-MLD-000. Instructions for Continued Airworthiness (Supplement), ref: 40186-02-A-MLD-000. Instructions for Continued Airworthiness, Bombardier Inc. document No. MSL39123002, Revision NC, dated 10 June 2019. Prior to installation of this change/repair it must be determined that the technical interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AERODATA AG</td>
<td>Installation of an Aerodata automatic flight inspection system(AD-AFIS-220MD).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Master Document List, document(AD-AFIS-220). Rev. 0 dated 13 September 2019. Operation in accordance with Aircraft Flight Manual Supplement, document(AFMT-171). Rev. 3 dated 11.09.2019. Instructions for Continued Airworthiness in accordance with the Aircraft Maintenance Manual Suppl...</td>
<td>Limited to B300 aircraft with installed Rockwell-Collins ProLine FusionAvionics. Prior to installation of this change/repair, it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the aircraft.</td>
<td>13/02/2019</td>
<td>Active</td>
<td>10070502</td>
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<tr>
<td>INGENIEURBUER O CALSBACH UG</td>
<td>Connector replacement - Service Bulletin SB STC 10070970-1</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval without any impact...</td>
<td>Service Bulletin SB STC 10070970-1</td>
<td>Applicable only for STC 1007</td>
<td>20/10/2022</td>
<td>Active</td>
<td>10070970</td>
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<tr>
<td>AVIONICS INTEGRATION &amp; ENGINEERING COorporation AG (AIEC)</td>
<td>Replacement of Laseref I by Laseref IV. Removed: Triple Honeywell Laserf I (2 x HG107SABMK and 1 x HG107SABSL). Installed: Triple Honeywell Laserf IV (HG2001G4) - Triple Shadin D/Converter (834510-00) - Removed: Triple Honeywell Laseref I.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List: AC5754-048, Rev. A for Continued Airworthiness. AC5754-085, Rev. A for Additional Airworthiness. AC5754-071, Rev. A for other revisions of the above listed documents approved by EASA.</td>
<td>S/N 5042 (HB-KG) Prior to installation of this change/repair, it must be determined that the relationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/09/2019</td>
<td>Active</td>
<td>10070808</td>
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<td>DOA AESGLOBAL LIMITED</td>
<td>INTRODUCTION OF ZERO PASSenger CONFIGURATION: Revision 1 extends the applicability Rev 2. Transfer from AES Ltd. to new STC holder AES Global Ltd.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>AES Limited Master Data List No. AES-009-97288MDL. Issue 4A5Limited Flight Manuals Supplementation AFM Supplement No. 89 Issue 1 FLIGHT MANUAL SUPPLEMENT - Introduction of...</td>
<td>Prior to installation of this change/repair, it must be determined that the relationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/04/2021</td>
<td>Active</td>
<td>10070986</td>
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<tr>
<td>LUFFTHANS TECHNIK AG</td>
<td>Replacement of existing Multi-Mode Receivers with HoneywellIntegratedMulti-Mode Receivers introducing S2AS capability</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental Type Design Document STD004-19/04FM Issue 1. Instructions for Continued Airworthiness A-00-19/73-ICA Issue/Rev. 02 Airplane Flight Manual Supplement A-00-19/70-AFM Issue/Rev. 01/005 later revisions of the above listed document(s) are approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair, it must be determined that the relationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/02/2019</td>
<td>Active</td>
<td>10071002</td>
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<tr>
<td>SABFINA TECHNICS BGC</td>
<td>Privacy door installation between the L11 lavatory and the crew rest.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List, ref 403338-01-B-MLA-R00 - Instructions for Continued Airworthiness, ref 403338-01-A-ICA-R01 - Master Minimum Equipment List supplement, ref 40339-01-A-MMELS-R00 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The Change related to this STC has to be embodied on aircraft with a minimum MOD 40416-01 - A330-200 - Electrical provision installation for the privacy door installation along the cockpit aisle. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/06/2019</td>
<td>Active</td>
<td>10071037</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10071041</td>
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<td>747-8</td>
<td>Installation of J-Music System. This change includes the required modifications to the reconfiguration and installation of all systems components for a fully functioning system.</td>
<td>Certification basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 23.1, CS 23.3, CS 23.21, CS 23.29, CS 23.49, CS 23.63, CS 23.143, CS 23.201, CS 23.203, CS 23.207, CS 23.251, CS 23.301,..</td>
<td>- Application of ORATEX covering and coating system installed without stitching. Rev. 1 includes &quot;on-condition&quot; NDT evaluation for the lifetime of the system. Rev. 2 extends the applicability.</td>
<td>23/09/2019</td>
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<tr>
<td>10071050</td>
<td>REV. 1</td>
<td></td>
<td>DORNIER 228-212</td>
<td>Installation of equipment for Implementation of full Mode SELS, EHXS and ADS-B Out capability on Dornier Do-228 aircraft.</td>
<td>Certification basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-ACNS Initial Issue.</td>
<td>Installation in accordance with RUAG Master Document List (MDL) no. MDL-8005685 Rev. B, dated 18.09.2019.</td>
<td>02/02/2021</td>
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<td>10071055</td>
<td>REV. 2</td>
<td></td>
<td>LANTZ-PRENA FOLIEN FACTORY GMBH</td>
<td>Application of ORATEX covering and coating system installed without stitching. Rev. 1 includes &quot;on-condition&quot; NDT evaluation for the lifetime of the system. Rev. 2 extends the applicability.</td>
<td>Certification basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 23.1, CS 23.3, CS 23.21, CS 23.29, CS 23.49, CS 23.63, CS 23.143, CS 23.201, CS 23.203, CS 23.207, CS 23.251, CS 23.301,..</td>
<td>- ORATEX System with unstitched integration on DR-series. Doc. No.ADC-S1-C0-02 Issue B dated 14th June 2019. Application of ORATEX covering &amp; coating system, Doc. No.ADC-S1-C0-051, Issue M, dated 5th March 2020.</td>
<td>19/10/2020</td>
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<td>10071060</td>
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<td>JET AVIATION AG</td>
<td>Interior reconfiguration of the noise area on the main deck of Boeing 747-300, S/N 23070, originally approved under FAA STC No. SA3572SW, subsequently modified with CAGA STC No. RAE-052. The reconfiguration includes the installation of two newsingle seats (approved for TTOI), a desk with an extension.</td>
<td>Certification basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): D-O2 Isolated Compartments - Deviation(s): D-O5 Fire Handlist. The requirements for environmental protection and the associated certified noise and/or emissions remain applicable to this certificate/approval.</td>
<td>- Master Data List JBSL- M-01189MDL-01, Issue 02- Aircraft Flight Manual Supplement JBSL- M-01189AIPM-01, Issue 04 concerning the above listed document(s) approved.</td>
<td>Applicable for S/N 23070 only. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10071065</td>
<td>REV. 1</td>
<td></td>
<td>GROWING ENGINEERING PARTNERS GMBH</td>
<td>Installation of IFE Streaming installation and Activation of a WiFi (based) IFE streaming solution consisting of a Head End Server Unit, Wireless Access Points and a Crew Control Panel Rev. 1. Transfer of STC from SR Technics Switzerland to groWING Engineering Partners GmbH.</td>
<td>Certification basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Master Data List S.0A04.001926-MDL, Rev. 03- Instructions for Continued Airworthiness Supplement S.0A04.001926-ICA, Rev. 00 concerning the above listed document(s) approved.</td>
<td>Applicable for S/N 23070 only. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>AEROMETALS INC.</td>
<td>10071066</td>
<td>REV. 1</td>
<td>FAA STC SR02926LA</td>
<td>S-92A</td>
<td>Installation of Engine Inlet Barrier Filter SystemRevision 1 to correct the STC holder name.</td>
<td>The Certification Basis for the original product as amended the following paragraphs at a later amendment: 14CFR Part 29: 29.1318 (b) Amdt 27-53/9-1317 (c) Amdt 27-49/4 requirements for environmental protection and the associate...</td>
<td>Master Drawing G1002 MDL-1. Rev. “F” dated September 17, 2011. Instructions for Continued Airworthiness: 1055-ICA-1, Rev. “F” dated June 15, 2011.</td>
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<td>GLOBAL AEROSPACE DESIGN CORPORATION</td>
<td>10071068</td>
<td>REV. 1</td>
<td>FAA STC STD0298CH</td>
<td>SEE FAA AML 767-300</td>
<td>Certification of Automatic Dependent Surveillance - Broadcast (ADS-B) OUT System utilizing upgraded Rockwell Collins TPR-901 Mode STransponders paired with the aircraft's existing GPS Source listed on AML number ST042809CH. Revision 1 adds Boeing 737-800 to the applicability list.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraphs at a later amendment: CS26 amnd: 25-153S(ix)(c)CS26 and 25-153S(ix)(d)CS26. The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraphs at a later amendment: CS26 amnd: 25-153S(ix)(c)CS26 and 25-153S(ix)(d)CS26.</td>
<td>Global Aerospace Design Corp. Master Drawing List No. GR0343 MDL Revision C, dated May 16, 2018, or later FAA approved/revision: Airplane Flight Manual Supplement (AFMS) listed on AML number ST042968CH.</td>
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<td>airbus engineering GmbH</td>
<td>10071070</td>
<td>REV. 1</td>
<td>EASA STC No. 525A</td>
<td>SEE FAA AML 767-300</td>
<td>Upgrade of Avionic System to WAAS Upgrade of existing Garmin GPS 400/ GNSS330A to GPS 400W / GNSS330W. This modification enables the WAAS functionality.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval and is complemented by EASA ACSN Amendment 2.</td>
<td>Modification in accordance with Engineering Work Order EW-2277-340000, Issue 2 and Description of Modification Issue 3.</td>
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<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>10071071</td>
<td>REV. 1</td>
<td>A330-343/341</td>
<td>A330-321/320/321/341-342</td>
<td>Installation of 90-Day Underwater Locator BeaconRev 1. Transfer of STC from GVH Aerospace Ltd - United Kingdom, to GVHAerospace (Ireland) Limited</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraphs at a later amendment: CS 25 at amendment 23.25.1, 25.26, 25.31, 25.305, 25.307, 25.321, 25.337, 25.341, 25.601, 25.651, 25.853, 25.1301...</td>
<td>- GVH Aerospace Instructions for Continued Airworthiness (ICA)GVH-10367-01 ICA-01 Issue 1- GVH Master Data List No. GVH-10357-01 MDL-1 Issue 1 - Instructions for continued airworthiness in accord...</td>
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<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>10071097</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>EASA Validation of FAA Approved AML Supplemental Type Certificate (STC)No. ST02936LA that installs RTCA DO 296B Version 2 ADS-B Out (ACSS Model transponder XS-650). This STC optionally installs the ACS5 XG-600 GPS/ISA the ADS-B Out positionsource and a Fail indicator and/or AT/CA225 Control panel as the...</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraphs at a later amendment: See EASA Approved ModelListThe requirements for environmental protection and the associated certified noise and/or emission...</td>
<td>- Master Data List (MDL), ACSS Doc Number 8809940-001 Rev. - dated 07 December 2018. Aircraft Flight Manual Supplement (AFMS), ACSS Doc Number 8809940-000Rev. - dated 07 December 2018.</td>
<td>None</td>
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<td>ROCKWELL COLLINS, Inc.</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
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**List of EASA Supplemental Type Certificates**

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<td>GDC ENGINEERING</td>
<td>03/07/2019 Active</td>
<td>The Certification Basis for the original product remains applicable for this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Certification Documents List CCN 8744B Issue 01/2020 Instructions for Continued Airworthiness 8744B-ICA-00 Issue 00/14 Master Certification Documents List CCN 8744B Issue 01/2020 Instructions for Continued Airworthiness 8744B-ICA-00 Issue 00/14 later revisions of the above listed document(s) approved/accepted. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10071127</td>
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<td>MICRO AERODYNAMICS, INC.</td>
<td>04/10/2019 Active</td>
<td>The Certification Basis for the original product remains applicable for this certification approval, except where amended by additional or later amendments if indicated on FAA STC SA12952. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Data List CTG MOD5004-320 documentation revision B dated 07 Oct 2019. AFM Supplement CTG MOD5004-1 Rev. Aor later revisions of the above listed document(s) approved/accepted under the EASA system. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>COMTECH GLOBAL AVIATION SERVICES S.A.</td>
<td>08/10/2019 Active</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Paragraph CS 25.603 at amendment 19/paragraphs. CS 25.301, 25.581, 25.609, 25.787, 25.831, 25.863, 25.869 at amendment 19/paragraphs. CS 25.811, 25.1301, 25.1411, 25.1701, 25...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>SCANDINAVIAN AVIONICS DESIGN A/S</td>
<td>07/10/2019 Active</td>
<td>CS-23 at amendment 3, the requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Installation in accordance with Scandinavian Avionics Master Document (MD) ref 211784; DOC; Rev 1, dated 04.10.2019 and Engineering Order ref 211784-EO, Rev 1, dated 03.10.2019. Operation in accordance with Scandinavian Avionics Aircraft Flight Manual Supplement (AFMS) ref 211784-AM A, Rev 1, dated 03.10.2019.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>AERO-DIENST GMBH</td>
<td>07/10/2019 Active</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: 25.581, 25.609, 25.809/25.1209 from CS-26 initial issue. The requirements for environmental protection and the associated.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>H4 AEROSPACE (UK) Ltd.</td>
<td>07/10/2019 Active</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition(s): CRI D-01 Installation of mini-suite type seating CRI D-02 Installation of seatbelt. The requirements for environmental protection and the associated.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>SIRIUM AEROTECH S.L.</td>
<td>REV. 4</td>
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<td>ATR 72-212 A</td>
<td>10071191</td>
<td>AGS-B Out upgrade for ATR 42/72 Rev. 1: addition of an alternate GPS antenna installation option Rev. 2: added reference to minor change SA-2033M for MMEL SRev. 3: Correction to include missing references to approved supplement</td>
<td>01/05/2023</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>A320-323/-342/-343</td>
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<td>A320-323/-342/-321</td>
<td>10071161</td>
<td>Activation of the Wireless IFES modification activates a Wireless (WiFi) Infotainment (Cabin Distribution) System on Airbus A330 aircraft. The full provision is installed in a separated modification. This Wireless IFES system is fed by the data connection via a Satcom system, which is installed th...</td>
<td>08/10/2019</td>
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<td>LEARJET Inc.</td>
<td>FAA STO398B0</td>
<td>CL-600-2B16 (604 VARIANT)</td>
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<td>Installation of a cockpit speaker mute switch.</td>
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<td>10071191</td>
<td>SABENA TECHNICS BGC</td>
<td>A319-132</td>
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<td>Re-installation of Litho Lift Cargo Loading System This STC re-installs the Litho Lift Cargo Loading System configuration (replacing the interim, bulk cargo, configuration).</td>
<td>17/10/2019</td>
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<td>10071193</td>
<td>STANDARD AERO</td>
<td>FAA STC ST0428CH-D</td>
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<td>HAWKER 850XP</td>
<td>10071193</td>
<td>Installation of a Collins IFIS-5000 Integrated Flight Information System</td>
<td>14/10/2019</td>
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GULFSTREAM AEROSPACE CORPORATION

10071226

STC Holder: GULFSTREAM

Active

Limitations/Conditions

The Certification Basis for the original product as amends the associated technical and environmental protection requirements. The requirements for the original product as amended by the associated technical and environmental protection requirements must be determined by the FAA.

Status

21/09/2020

The Certification Basis for the original product is unaltered and remains applicable to the original product as amended.

GULFSTREAM AEROSPACE CORPORATION

10071227

STC Holder: GULFSTREAM AEROSPACE CORPORATION

Active

Limitations/Conditions

The Certification Basis for the original product as amended.

Status

15/10/2019

The Certification Basis for the original product as amended remains unaltered and applicable to the original product as amended.

GULFSTREAM AEROSPACE CORPORATION

10071229

STC Holder: GULFSTREAM AEROSPACE CORPORATION

Active

Limitations/Conditions

The Certification Basis for the original product as amended.

Status

17/10/2019

The Certification Basis for the original product as amended remains unaltered and applicable to the original product as amended.

Cessna Dueseldorf Citation Servicing Center GmbH

10071261

STC Holder: CESSNA

Active

Limitations/Conditions

The Certification Basis for the original product as amended.

Status

24/03/2020

The Certification Basis for the original product as amended remains unaltered and applicable to the original product as amended.

Cessna Dueseldorf Citation Servicing Center GmbH

10071262

STC Holder: CESSNA

Active

Limitations/Conditions

The Certification Basis for the original product as amended.

Status

16/10/2019

The Certification Basis for the original product as amended remains unaltered and applicable to the original product as amended.
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<td>ARIFILM CAMERA SYSTEMS, INC</td>
<td>FAA-SR02695LA</td>
<td>130 B4/130 T2</td>
<td>Installation of Airfilm Utility Mount</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments/alterations if indicated on FAA STC. The amendments for the associated certified noise and/or emissions levels of the original product...</td>
<td>1. FAA amended the Certification Basis for the original product in the FAA STC ST01929WI. The amendments for the associated certified noise and/or emissions levels of the original product...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the aircraft product.</td>
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<td>SWS CERTIFICATION SERVICES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>777-200, 777-300ER</td>
<td>RE-PITCH OF OUTBOARD ECONOMY CLASS SEATS</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for the associated certified noise and/or emissions levels of the original product...</td>
<td>SWFS Certification Services Ltd Master Configuration Scheme No 3199-MCS-01 Issue 1.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the aircraft product.</td>
<td>18/10/2019</td>
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<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>REV. 2</td>
<td>AW1313, AW1319, AW1319</td>
<td>Installation of AeroStretcher Mk IV Modular EMS System in accordance with Master Data List G16.4386-01-MDL-01 Initial Installation of AeroStretcher Mk IV Modular EMS System Limited to ...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for the associated certified noise and/or emissions levels of the original product...</td>
<td>Flight Manual Supplement G16.4386-01 FMS-01 Issue 2011/02-01-00</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the aircraft product.</td>
<td>22/12/2020</td>
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<td>JET AVIATION AG</td>
<td>REV. 2</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>WLAN Activation/Demonstration of aircraft immunity to IEEE 802.11abg/n/ ac WLAN transceivers conform to IEEE 802.11abg/n/aasea.europa.eu</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: C5-25 Amnd 23: 769, 831.1, 831.11, 1301, 1309, 1351, 1353, 1357, 1360, 1341, 1501, 1529, 1469, 1581, 1585, 1707. The requirement...</td>
<td>Revision 2-Cert Plan: JBSL-M-01351/ALF-01 Issue 2011/02-01-00 as detailed in SWSCertification Services Ltd Design Report 2011/02-01-00</td>
<td>As per AFM JBSL-M-01351ADF-01. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the aircraft product.</td>
<td>3/10/2022</td>
<td>Active</td>
<td>10071326</td>
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<tr>
<td>BLACKHAWK AEROSPACE TECHNOLOGIES IN</td>
<td>FAA STC TST01929WI</td>
<td>500/550/560/5550</td>
<td>Installation of Garmin G700 TXi EFIS System and GWX 75 Weather Radar</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments/alterations/linifications if indicated on FAA STC TST01929WI. The requirements for the associated certified noise and/or emissions levels of the original product...</td>
<td>1. FAA amended the Certification Basis for the original product in the FAA STC TST01929WI. The requirements for the associated certified noise and/or emissions levels of the original product...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change/repair will introduce no adverse effect upon the airworthiness of the aircraft product.</td>
<td>20/10/2019</td>
<td>Active</td>
<td>10071381</td>
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<td>STC Holder</td>
<td>Garmin GFC700 Autopilot endorsementEASA STC 10048730 approves the installation of TechniCentur 2.0fLowed Engines on the Diamon DA 42. Compared to the OEM approvedTechniCentur option, the 2.05 provides more engine power and in the structure of STC 1048730, just the KAP140 Autopilot System h...</td>
<td>Da 42</td>
<td>GB</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10071407</td>
<td>STC Holder</td>
<td>Garmin GFC700 Autopilot endorsementEASA STC 10048730 approves the installation of TechniCentur 2.0fLowed Engines on the Diamon DA 42. Compared to the OEM approvedTechniCentur option, the 2.05 provides more engine power and in the structure of STC 1048730, just the KAP140 Autopilot System h...</td>
<td>Da 42</td>
<td>GB</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10071421</td>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>Installation of GPS antenna and WAAS GPS receiver</td>
<td>650</td>
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<tr>
<td>10071441</td>
<td>ACC AIRCRAFT CABIN COMPONENTS Gmbh</td>
<td>Change to EMBRAER 190-100LR Cabin Refurbishment, Replacement Seat Cover of All Exit Row ZODIAC Seats</td>
<td>EUR 190-100 LR</td>
<td></td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associatedcertified noise and emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>Continued Airworthiness for Installation of GPS/WAAS Receiver Doc Ref:210697-CA, Issue 1 dated 20 September 2019 or later revisions of the above listed document(s) approved/accepte by the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/10/2019</td>
<td>Active</td>
<td>10071421</td>
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<tr>
<td>10071445</td>
<td>JET AVIATION AG Rev. 3</td>
<td>KABand Satcom System InstallationThis project is installing a Honeywell Jet Wave MCS-8200 KABand Satcom system on the rear fuselageRev. 1: inclusion of new installation location for the A318 and A320series and inclusion of A318 installationRev. 2: Typo and editorial changesRev. 3: Correction of appendices...</td>
<td></td>
<td></td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associatedcertified noise and emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>As defined in ACC Maintenance Manual Supplement ref. MAS1041-01 Issue A this Major Change Approval (MCA) is limited to an operation time of 1 (one) year. After that period the MCA shall be removed and the aircraft shall be returned into its previously approved configuration.</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/11/2019</td>
<td>Active</td>
<td>10071442</td>
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<td>10071446</td>
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<td>A320-252N-271N-272N-273N A320-2323-232-233-231N A320-252N-271A-212-214-215 A319-212-213-212-151N A319-111-212-113-114-115 A318-111-212-121-122 A320-252N-271A-212-214-215 A319-212-213-212-151N A319-111-212-113-114-115 A318-111-212-121-122</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associatedcertified noise and emissions levels of the original product unaltered and remain applicable to this certificate/ approval.</td>
<td>As defined in ACC Maintenance Manual Supplement ref. MAS1041-01 Issue A this Major Change Approval (MCA) is limited to an operation time of 1 (one) year. After that period the MCA shall be removed and the aircraft shall be returned into its previously approved configuration.</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed design change, repair and/or replacement will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/01/2021</td>
<td>Active</td>
<td>10071445</td>
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### List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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</thead>
<tbody>
<tr>
<td><strong>10071447</strong></td>
<td>STC Holder</td>
<td>Certification basis</td>
<td>EMBRAER 10071509</td>
<td>PATS installation of VIP Aircraft Interior</td>
<td>- PATS Aircraft, LLC Master Drawing List 6056-0001-MDL, revision F dated 12 February 2015 - Airworthiness Limitations Section (section 4 of the MPDS Report) TM-6950-MPDS</td>
<td>30/10/2019</td>
<td>Active</td>
</tr>
<tr>
<td><strong>10071509</strong></td>
<td>STC holder</td>
<td>Certification basis</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>A300-600 ST Bajaka - EFB arm mount</td>
<td>The change consists of the installation of hardware and power supply enabling windows for the installation of Panasonic CF-20 Electronic Flight Bag (EFB). This change installs the adaptation supports, the PFS, the AGG sliding of Mount and associated cradle on CAP7...</td>
<td>04/11/2019</td>
<td>Active</td>
</tr>
<tr>
<td><strong>10071512</strong></td>
<td>STC holder</td>
<td>Certification basis</td>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY SNC</td>
<td>Installation of Rockwell Collins Automatic Direction Finder</td>
<td>This STC was originally issued as a validation of Brazilian CTS (STC) no. 2009/950-07 (31 May 2009) and transferred to a United States based Master Approval Holder (DAH) on 17 July 2019.</td>
<td>25/11/2019</td>
<td>Active</td>
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<tr>
<td><strong>10071527</strong></td>
<td>STC holder</td>
<td>Certification basis</td>
<td>SABENA TECHNICS BGC</td>
<td>ADS-B OUT System Installation</td>
<td>ADS-B OUT System Installation on ATR42/72/500 (Supplement): 0724-02-A-MDL-R04</td>
<td>25/04/2022</td>
<td>Active</td>
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<tr>
<td><strong>10071538</strong></td>
<td>STC holder</td>
<td>Certification basis</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>GTX 3000 Dual Transponder Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remained unchanged and remain applicable to this certificate/ approval.</td>
<td>None</td>
<td>05/11/2019</td>
</tr>
<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA Code</td>
<td>Movie(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>DC AEROSPACE SOLUTIONS, LLC</td>
<td>ST02562A</td>
<td>FAA ST</td>
<td>A321-272KN X-272N X-272</td>
<td>Installation of Flight Deck Thermal Acoustic Insulation Blanket</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC 0874-02-A-ICA-R001.</td>
<td>- Master Drawing List, Document Number MDL-2-194190-001, Rev. B dated 17 September 2019</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>SABENA TECHNICS BGC</td>
<td>ST01505</td>
<td>FAA ST</td>
<td>A320-233, A320-232, A320-231</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: 25.305(a) “Vibration analysis of critical areas” and 25.307(a) “Noise of critical areas”</td>
<td>- Master Data List, ref 0874-02-A-MLD-R03-Instructions for Continued Airworthiness, ref. 0874-02-A-ICA-R001-Aircraft Flight Manual Supplement, ref. 0874-02-A-AMF-R00</td>
<td></td>
<td>This modification is compatible with Extended Service Goal as described in section 3.2 of Instructions for Continued Airworthiness (Supplement), ref. 0874-02-A-ICA-R001 Models not already fitted with ARINC 791 provisions for the installation of a dorsal large Satellite Communication Antenna between frames 18 February 2016 or later FAA approved revisions.</td>
</tr>
<tr>
<td>ALPHA AVIATION, INC.</td>
<td>ST01505</td>
<td>FAA AML</td>
<td>Installation of a Three Point Restraint System</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA2015CH. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>Alpha Aviation, Inc. Installation and Maintenance Manual dated 23 December 2015, or later FAA approved revisions. The Three Point Torso Restraint System installation is to be inspected and maintained in accordance with the FAA STC.</td>
<td></td>
<td></td>
</tr>
<tr>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td>ST01505</td>
<td>FAA ST</td>
<td>737-800</td>
<td>Alternative Maximum Takeoff Weight Reduction of the certified maximum take-off weight of the aircraft to 17900 kg in accordance with EASA record 12/2017.</td>
<td>The Certification Basis (CB) for the original product remains applicable to the certificate/approval. This certificate/approval involves a change to the requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>Airplane Flight Manual Supplement MODE0287_AFMS1 rev 3or later revisions of the above listed document(s) approved/updatedunder the EASA system.</td>
<td>Applicable to Boeing 737-800 aeroplanes, equipped with winglets and CFM56-7B26E engines. APM option code 9986951-H Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherwise installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>STC Holder</td>
<td>System</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10071581</td>
<td>ALPHA AVIATION, INC.</td>
<td>FAA SA01880CH</td>
<td>Installation of a Three Point Tono Resistant System.</td>
<td>1. Alpha Aviation Inc. Installation and Maintenance Manual as listed on AML SA01880CH, reissued 20 December 2016, or later FAA Approvals revisions 2. Approved Model List (AML) No. SA01880CH, reissued 20 December 2016, or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that there is no interrelationship between this design change and any other previously installed design change and/or repair will introduce roadways effect upon the aeroworthiness of the product. The STC is valid for those aircraft types and model...</td>
<td>07/11/2019</td>
<td>Active</td>
</tr>
<tr>
<td>10071587</td>
<td>MORSON PROJECTS LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>MT-PROPELLER</td>
<td>Installation, Operation and Maintenance according to MorsonProjectsLimited Change Report MPX030A= ( Issue 1, dated 23 October 2019) (and/or later revisions of the above document(s)) approved/accepted under the EASA system.</td>
<td>Limited to the following serial numbers of aircraft that are fitted with Rockwell Collins Proline 21 Avionics with dual Rockwell Collins TDR-940 transponders compliant with RTCA DO-206B and Rockwell Collins GPS-4009GSN position sensor qualified to EASA ETRO-C145 (and associated SBA/Squalified GNSS an...</td>
<td>08/11/2019</td>
<td>Active</td>
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<tr>
<td>10071603</td>
<td>J &amp; C ENGINEERING SERVICES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>Installation in accordance with FAA Approved Aircraft Flight Manual/Supplemental document number EGR-887-...</td>
<td>None</td>
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<td>08/11/2019</td>
<td>Active</td>
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<tr>
<td>10071611</td>
<td>LIFEPORT, LLC</td>
<td>FAA ST04307CH</td>
<td>Installation of a Medical Interior.</td>
<td>This STC requires prior endorsement by Textron of the modificationslisted in the Medical Interior Collector document number 7490007, Rev... If the holder agrees to permit another person to use this certificate the product, the holder must give the other person written evidence of that permission...</td>
<td></td>
<td>08/11/2019</td>
<td>Active</td>
</tr>
<tr>
<td>10071662</td>
<td>AIRBUS S.A.S.</td>
<td>A330-743L</td>
<td>Airbus Interiors Services (AIS) MOD CJ1970 Courier Area Installation sucks A330-743L Beluga XL This Courier Area MOD1970 provides a module integrated in the A330-743LBeluga XL aircraft, between the cockpit and the aft pressurized bulkhead. This module provides transportation provisions up to four...</td>
<td></td>
<td></td>
<td>11/11/2019</td>
<td>Active</td>
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<tr>
<td>10071663</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>REV. 2</td>
<td>Installation of MTV-6 (and/or) MT-6196-B) on DA42 series.</td>
<td></td>
<td></td>
<td>28/04/2023</td>
<td>Active</td>
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<td>STC Holder</td>
<td>FAA STC No.</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>AERODATA AG</td>
<td>10071634</td>
<td>REV. 1</td>
<td></td>
<td>B300</td>
<td>Installation of an Aerodata automatic flight inspection system (AD- AFIS-1012DF).</td>
<td>The Certification Basis for the original product as amended by following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: Elcetred to comply with CS-23 Amdt. 4 except for 23.51F and 23.785 at FAR-23 Amdt. 34. The requirements for environmental pro...</td>
<td>Installation in accordance with Master Document List, document MLD-1/C-303 Rev. 0 dated 01 October 2019</td>
</tr>
<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>10071643</td>
<td></td>
<td>SEE FAA AML</td>
<td>B300</td>
<td>Modification of Rockwell Collins VHF COMM receiver installation Rev: the original approved configurations (by Italian ENAC and Ireland CAA) were complemented by additions of the Boeing models 737-300/7F 737-800, 737-900, 747-400, 747-400F, 777-200, 777-300, 777-200LR, 777-300ER, DC-...</td>
<td>Requirements at a later amendment JAR 25 Amdt 15-25.601(5-60)(a)(25.612(b)(25.1301)(a)(25.1303)(a)(25.1519)</td>
<td>Rockwell Collins Drawing Index 502-4250-001, Rev. T, dated 27 January 2012, or later FAA approved revision.</td>
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<tr>
<td>SPAES GmbH &amp; Co. KG</td>
<td>10071645</td>
<td></td>
<td>SA 365 N1</td>
<td>SA 365 N</td>
<td>Installation of automatic deployable ELT</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Document List MLD-365-245-25-01, Issue 00- Instructions for Continued Airworthiness ICA-365-245-25-01, Issue 00- Flight Manual Supplement FMS- SPAES-0045, Issue 00</td>
</tr>
<tr>
<td>JLC AERONAUTICAL LLC</td>
<td>10071650</td>
<td></td>
<td>FAA STC SAA6225W</td>
<td>2. DHG-6 SERIES 400</td>
<td>Installation of JLC Aeronautical Vapour Cycle Air Conditioner This is a revision to the French STC FM-02-01 dated 15 February 2001.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Drawing List MA-001, Rev. 4 dated 6 December 2012 or later FAA approved revision.2. FAA approved revision.</td>
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<tr>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY INC</td>
<td>10071654</td>
<td></td>
<td>FAA STC SAA8483AT</td>
<td>EMB-505</td>
<td>Installation of a Tire Pressure Sensing Kit</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Data List (MDL) 17008-001, Rev. --, dated 21 November 2018, or later FAA approved revision.2. FAA approved revision.</td>
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<tr>
<td>CAVOK</td>
<td>10071663</td>
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<td></td>
<td>B300</td>
<td>Installation of forward pressure bulkhead cable feedthrough.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation in accordance with Installation Instructions document number 350-084-350-04-00, Issue B, dated 25/09/2019</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>B. DHC-6 SERIES 400</td>
<td>Installation of Replacement Fuel Manifold Tubes</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation in accordance with Transport Canada Civil Aviation (TCC)Approved Rocky Mountain Aircraft (RNA) Modification Instruction E108610.Revision B, dated March 11 2016 or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/11/2019</td>
<td>Active</td>
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<tr>
<td>BOEING IRELAND LIMITED</td>
<td>737-800</td>
<td>Zero Passenger LOPA for B737 This Change removes all existing passenger seats from the passenger cabin of the aircraft. Operation of the aircraft is restricted livelihood with flight attendants crew only (no passengers allowed).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Change / Repair Approval BIL-00107 APP Revision 06, dated 14th Nov 2019Aircraft Flight Manual Supplement BIL-00107- AFM01 Revision 01Manual Supplement BIL-00107- MS51 Revision 00or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/11/2019</td>
<td>Active</td>
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<tr>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>737-700</td>
<td>Activation of Wireless Local Area Network (WLAN) for the purpose of passenger connectivity.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): F-108 Security Protection of Aircraft Systems and Networks: the following paragraph(s) at alater amendment. The following paragraphs of CS-25, as of Aman...</td>
<td>The change is defined with the following revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/11/2019</td>
<td>Active</td>
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<tr>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>737-700</td>
<td>Installation of Inmarsat Satcom System: Installation/Installation of an Inmarsat antenna mounted on upper part of the fuselage (between STA 460 and STA 480) and a Data Link Unit (DLU).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): F-108, Security Protection of Aircraft Systems and Networks: the requirements for environmental protection and the associated certificate noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>The change is defined with the following revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/11/2019</td>
<td>Active</td>
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<tr>
<td>CANARD AEROSPACE CORPORATION</td>
<td>737-800-700-800</td>
<td>Installation of ADS-B Qualified Transponders and new Integrated Multi-Mode Receivers (IMMRs) on Boeing 737</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS-25 and/or 4-25.200(b), 25.391(a),(b), 25.303(c), 25.356(a),(b), 25.370(a), 25.321(4),(5), 25.335(a),(b), 25.961(c); 25.571(c),(d), 25.651, 25.623; 25.655; 25.605; 25.607; 25.611;...</td>
<td>- Canard Aerospace Corporation (CAC) Document Drawing J6216, Revision N dated 8 March 2021, or later FAA approved revision - Airplane Flight Manual Supplement, Canard Aerospace Document Number#94-3628-03, Revision NIC, FAA approved 8 July 2019, or later FAA approved revision - Instructions for Continued Airworthiness: Document J6216.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/06/2021</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10071713</td>
<td>REV 1</td>
<td>FAA STC ST0680LA-D</td>
<td>GIV, S/N 1563 ONLY</td>
<td>Installation of passenger cabin interior</td>
<td>The Certification Basis for the original product remains applicable if certificated approval, except where amended by additional or later amendments if included on FAA STC ST0680LA-D. The requirements for environmental protection and the associated denoted noise and/or emissions levels of the...</td>
<td>Gulfstream Aerospace Corporation Index List II 11T01555, Revision E dated 17 July 1999, or later FAA approved revision.</td>
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<td>LUFFTHansa TECHNIN AG</td>
<td>10071716</td>
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<td>A340-341-242/442-443</td>
<td>A320-211-212-213</td>
<td>Installation of a Panasonic In-Flight CVR-120A Cockpit Voice Recorder.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements (new paragraph(s) at a later amendment: CS25.029(a)(2); 25.302(b); 25.303, 25.561(c); 25.601, 25.569(e); 25.698(b); 25.1301(a)(1); 25.1301(a)(2); 25.1301(a)....</td>
<td>- Master Data List STD001-1909M Issue Revision 1.0.</td>
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<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>10071721</td>
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<td>FAA STC ST04387AT</td>
<td>A320-271N, -272N, -273N</td>
<td>Class Divier Installation in an Airbus A320 Series aircraft</td>
<td>The Certification Basis for the original product remains applicable if certificated approval, except where amended by additional or later amendments if included on FAA STC ST04387AT. The requirements for environmental protection and the associated denoted noise and/or emissions levels of the...</td>
<td>1. HAECO Cabin Solutions, LLC Master Data List 19T011.001, Revision IR dated 8 November 2019, or later FAA approved revisions. 2. HAECO Cabin Solutions Instructions for Continued Airworthiness 19T011R005, Revision 00, dated 17 October 2019, or later FAA acceptance/revision.</td>
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<td>LEARJET inc.</td>
<td>10071724</td>
<td></td>
<td>FAA STC ST0027LA-D</td>
<td>31A</td>
<td>Installation of the Universal Avionics Systems Corporation Dual Bare Air...</td>
<td>The Certification Basis for the original product remains applicable if certificated approval, except where amended by additional or later amendments if included on FAA STC ST0027LA-D. The requirements for environmental protection and the associated denoted noise and/or emissions levels of the...</td>
<td>1. Learjet Master Data List BAS31010402, Revision &quot;C&quot;, dated 24 June 2011, or later FAA Approved revision. The following documents are required for MDL BAS31010402, - Configuration of this STC, C-00102, Fitment Flight Management System Supplement BAS31040208, Revision NC, FAA Approved on 24 June 2011, or later FAA Approv-.</td>
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<td>WEST STAR AVIATION, LLC/ bia WEST STAR AVIATION</td>
<td>10071725</td>
<td></td>
<td>MYSTÈRE FALCON 900 (MF900)</td>
<td>FALCON 900EX (F900EX)</td>
<td>Installation of Universal CVR-120A/Installation of a Universal CVR-120A Cockpit Voice Recorder</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (sc): CRI F-9122-003 Flight Recorder and Data Link Recording CRI F9128-03 Non-Rechargeable Lithium battery installation. EASA Certification Basis (Conti...</td>
<td>- MDL 18336L-1736 Revision B- AFM 18336A-1736 Rev B- AFI-ICA 18336P-1760 Rev R for later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>10071727</td>
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<td>FAA STC ST00549WY</td>
<td>777-300ER</td>
<td>Installation of a Panasonic In-Flight Entertainment System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements (new paragraph(s) at a later amendment: CS25, 25 at Armit, 4 §25.631, §25.632, §25.635, §25.636, §25.637, §25.1335, §25.1437 These requirements for environmental...</td>
<td>Delta Engineering Master Data List 0143-00432-2177, Revision E, dated 27 August 2019. The Instructions for Continued Airworthiness as listed on the Master Data List are required within installation.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
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<td>10071729</td>
<td>LORD CORPORATION</td>
<td>212 205B 205A-1 205A 205B</td>
<td></td>
<td>737-800</td>
<td>Part Manufacturer Approval (PMA) LORD Tension StrapNo. PDA84663 (supplement No. 135 for installation of LORD Tension StrapAssembly (PMA) LORD204-310-101101) on the Main Rotor.</td>
<td>The Certification Basis for the original product as amended by the following paragraphs(s) at a later amendment: 1) As no compliance demonstration with the certification basis for OSDconstituents CCD and FCD has been performed, the aircraft is not allowed to be registered in any EASA member state without demonstration of compliance to CCD and FCD OSD requirements. 2) The aircraft is required to be retrofitted with an Enhanced ADS-B system, with Collins HSI-85 (ex-factory)</td>
<td>Instructions for Continued Airworthiness for LORD PI NLORD204-310-101101, dated 29 May 2019</td>
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<tr>
<td>10071731</td>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td></td>
<td>737-800</td>
<td>Alternative Maximum Takeoff (AMT) Weight Reduction of the certified maximum take-off weight of the aircraft to 69690 kg, and maximum landing weight to 65317 kg.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended and remain applicable to this certificate/approval.</td>
<td>Applicable to Boeing 737-800 aeroplanes, equipped with engines CFM56-7B engines, AFM option code 5986951</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10071732</td>
<td>AERO-DIENST GMBH</td>
<td>55</td>
<td></td>
<td>737-800</td>
<td>RNP-1 i.a. CS-ACNS Issue 02, Subpart C</td>
<td>The Certification Basis for the original product remains applicable to this certification/approval. This certificate/approval involves a change to the requirements for environmental protection and the associated certified noise and/or emissions levels.</td>
<td>Airplane Flight Manual Supplement Issue 2019- L-21-DG-106, Rev. 00(&lt;&gt;)</td>
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<tr>
<td>10071737</td>
<td>LUFTHANSA TECHNIC AG</td>
<td>A319-115</td>
<td></td>
<td>737-800</td>
<td>A319 - Cabin Modification/ The existing VIP cabin interior, which has been installed by Airbus shall be modified to install one new compartment and to modify several areas within the cabin. In addition some systems shall be modified and accommodate new in-flight Entertainment system will be installed.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - The following paragraph(s) at a later amendment: 29.092 from CS-25/initial issue - CS-ACMS- Issue 2. The requirements for environmental protection and the associated certified noise and/or emissions levels.</td>
<td>- Airplane Flight Manual Supplement Issue 2019- L-21-DG-106, Rev. 00(&lt;&gt;)</td>
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<tr>
<td>10071738</td>
<td>SIRIUM AEROTECH S.L.</td>
<td>REV. 1</td>
<td>737-500, 757-200-300/737-100C-200/-200C-300/-400</td>
<td>737-800</td>
<td>ADS-B System Upgrade for B737 and B757Revision 1; added reference to minor change SA02-40M for MMEL-S</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>MASTER DATA LIST SA19-035-MDL Rev 02B737 AMD SUPPLEMENT SA19-035-AFMS-737 Rev 02B737 AMD SUPPLEMENT SA19-035-AFMS-737 Rev 02B737 AMD SUPPLEMENT SA19-035-MMELS rev for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
</tr>
<tr>
<td>10071752</td>
<td>J &amp; C AEROSPACE DESIGN ANDCERTIFICATI ON SERVICES EUE Ltd.</td>
<td></td>
<td>737-800, 737-700, 737-800</td>
<td>Survival ELT Installation/Survival Emergency Locator Transmitter (ELT) ELTA AD24065 Series PNI013659020 Installation in cabin interior in enclosed overhead bin.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product as amended and remain applicable to this certificate/approval.</td>
<td>Master Data List doc JCAEROMDL0447 issue 1.0 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Certification basis</td>
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<td>System Record</td>
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<td>10071756</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>Installation of Garmin GPS 175 Navigation System, GNX 375 Navigation System with AG58-8 In/Out Transponder, and GNC 355 Navigation System with WHP COM Radio Rev. 1: Addition of F190, FA190, FA192, FRA150, FA16, PA168 and PC-7 to the Approved Model List (AML).</td>
<td>For the GPS 175, a GPS/SBAS antenna must be previously or concurrently installed. For the GNC 375, a GPS/SBAS antenna and transponder antenna must be previously or concurrently installed. For the GNC 395, a GPS/SBAS antenna and a VHF COM antenna must be previously or concurrently installed.</td>
<td>01/08/2023</td>
<td>Active</td>
<td>10071756</td>
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<tr>
<td>10071757</td>
<td>HAEKO CABIN SOLUTIONS, LLC</td>
<td>Class Divider Installation in a Boeing 787-8 aircraft</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/11/2019</td>
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<td>10071757</td>
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<td>10071764</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>CNC System - Activation The modification activates the new Complementary Ground Component (CGC) System as a part of the European Aviation Network (EAN).</td>
<td>Installation only on models that satisfy the criteria defined in configuration sheet AIXU01-53-005-01206-A1 Revision 1.</td>
<td>23/01/2020</td>
<td>Active</td>
<td>10071764</td>
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<tr>
<td>10071765</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>MSS Satcom system installation The modification installs the new MSS Satcom system installation.</td>
<td>The installation can be performed only on models that satisfy the criteria defined in configuration sheet AIXU01-53-005-01206-A1 Revision 1. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/11/2019</td>
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<td>10071765</td>
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<tr>
<td>10071779</td>
<td>AIRBASE CERTIFICATION ENTERPRISE VICES (ACES) INC.</td>
<td>Seat Table Inset Installation Seat Table Inset Installation in the middle seat of Triple Passenger Seat Assembly – forward facing – on Boeing model 737-8.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/11/2019</td>
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### List of EASA Supplemental Type Certificates

**EASA STC no.**

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<th>STC Holder</th>
<th>REV.</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
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<tr>
<td>Luftansa Technik AG</td>
<td>REV. 1</td>
<td>747-400</td>
<td></td>
<td>Ribbes Film on Lower Fuselage/ Application of self-adhesive film to lower fuselage aft of STC 440. Amiocrusture pattern is applied in the film in order to influence the boundary layer and reduce drag (Revision 1). Improved Ribbes film material T1TF2.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any...</td>
<td>- STC-AT-140 Survey Configuration. In the exterior of the helicopter, this STC combines the installation of: a) Rivièr VY-1 eye-safe laser scanner on the forward RH side of the helicopter (according to nonmodified STC EASA IM.9-5.0158, validating FAA SR074/AAL, a) SIVEO EOCR Camera on the forward LH side (...</td>
<td>16/06/2020</td>
<td>Active</td>
<td>10771802</td>
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<td>Aerotec &amp; Concept</td>
<td>REV. 2</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td></td>
<td>Aeroec UCS AML Stretch뢰 Revision of AMT STC 10771815 to address additional aircraft types: Boeing 737/ Boeing 787/ ATR 42/ATR72 / DHC-8 / EMB-145</td>
<td>The Certification Basis for the original product as amended/the following additional or alternative aeroworthiness requirements the following paragraph(s) at a later amendment: CS 25.301, 25.303,25.305, 25.307, 25.321, 25.331, 25.337, 25.341, 25.561, 25.613, 25.625,25.785, 25.789, 25.791, 25.803,...</td>
<td>Master Data List MDL- SEE EASA APPROVED MODEL LISTAFM Supplement / AFM5: SEE EASA APPROVED MODEL LIST/Instructions for Continued Airworthiness / ICA: SEE EASA APPROVED MOD085/Totaler revisions of the above listed document(s) approved/accepted the EASA system.</td>
<td>- Per AFM supplement: SEE EASA APPROVED MODEL LIST. Per ICAsupplement: SEE EASA APPROVED MODEL LIST. Installation is limited to configurations as per sharing ref 025-18113-101 or025-18113-102 or 025-18113-103 or 025-18113-105 without Medical or Oxygen Equipment.Prior to installa...</td>
<td>27/04/2021</td>
<td>Active</td>
<td>10771815</td>
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<tr>
<td>STC-AT-140 Survey Configuration</td>
<td>REV. 1</td>
<td>AS 350 B3 AS 350 B2</td>
<td>STC-AT-140 Survey Configuration. In the exterior of the helicopter, this STC combines the installation of a Rivièr VY-1 eye-safe laser scanner on the forward RH side of the helicopter (according to nonmodified STC EASA IM.9-5.0158, validating FAA SR074/AAL, a) SIVEO EOCR Camera on the forward LH side (...</td>
<td>The Certification Basis for the original product as amended/the following additional or alternative aeroworthiness requirements the following paragraph(s) at a later amendment: CS-27.1, 27.25, 27.27, 27.29, 27.45, 27.51, 27.65,27.87, 27.71, 27.73, 27.75, 27.79, 27.141, 27.143, 27.145, 27.146,...</td>
<td>Change Approval Sheet, doc. ref. STC-145-G-AS Rev.0 Flight Manual Supplement, doc. no. STC-140-G-ARM Rev. 1Maintenance Manual Supplement, doc. no. STC-140-G-ARM Rev. 1: For later revisions of the above listed document(s) approved/accepted the EASA system.</td>
<td>The following changes must be embedded prior to the installation of this STC: Airfilm Camera System STC ref. EASA IM.9-5.0158 (Validating FAA STC0971UHL/A)(B); Airfilm Camera System STC ref. FAA STC0971UHL/A(C); Airetch Aeroshunt STC ref. EASA 10016802 Validating TCCAS</td>
<td>29/11/2019</td>
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<td>STC-AT-140 Survey Configuration</td>
<td>REV. 1</td>
<td>AS 350 B3 AS 350 B2</td>
<td>STC-AT-140 Survey Configuration. In the exterior of the helicopter, this STC combines the installation of a Rivièr VY-1 eye-safe laser scanner on the forward RH side of the helicopter (according to nonmodified STC EASA IM.9-5.0158, validating FAA SR074/AAL, a) SIVEO EOCR Camera on the forward LH side (...</td>
<td>The Certification Basis for the original product as amended/the following additional or alternative aeroworthiness requirements the following paragraph(s) at a later amendment: CS-27.1, 27.25, 27.27, 27.29, 27.45, 27.51, 27.65,27.87, 27.71, 27.73, 27.75, 27.79, 27.141, 27.143, 27.145, 27.146,...</td>
<td>Change Approval Sheet, doc. ref. STC-145-G-AS Rev.0 Flight Manual Supplement, doc. no. STC-140-G-ARM Rev. 1Maintenance Manual Supplement, doc. no. STC-140-G-ARM Rev. 1: For later revisions of the above listed document(s) approved/accepted the EASA system.</td>
<td>The following changes must be embedded prior to the installation of this STC: Airfilm Camera System STC ref. EASA IM.9-5.0158 (Validating FAA STC0971UHL/A)(B); Airfilm Camera System STC ref. FAA STC0971UHL/A(C); Airetch Aeroshunt STC ref. EASA 10016802 Validating TCCAS</td>
<td>29/11/2019</td>
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<td><strong>Certification basis</strong></td>
<td><strong>Associated Technical Documentation</strong></td>
<td><strong>Limitations/Conditions</strong></td>
<td><strong>Date of Issue</strong></td>
<td><strong>Status</strong></td>
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<td><strong>BOEING IRELAND LIMITED</strong></td>
<td><strong>REV. 1</strong></td>
<td><strong>10071905</strong></td>
<td><strong>A380-341</strong></td>
<td><strong>Installation of a Panasonic Avionics In-Flight Entertainment (IFE) System</strong></td>
<td><strong>Boeing Technical Services UK Ltd Instructions for Continued Airworthiness (ICAI) B21792-ICA00 Revision B</strong></td>
<td><strong>Prior to installation of this change/repair it must be determined that the change/repair it must be introduced no adverse effect upon the airworthiness of the product.</strong></td>
<td><strong>04/12/2019</strong></td>
<td><strong>Active</strong></td>
<td><strong>10071905</strong></td>
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<tr>
<td><strong>DELTA ENGINEERING CORPORATION</strong></td>
<td><strong>REV. 1</strong></td>
<td><strong>10071909</strong></td>
<td><strong>JET AVIATION AG</strong></td>
<td><strong>EGPWS Upgrade/Removed</strong></td>
<td><strong>- Master Data List Doc. No. JBLS-M-0134MDL-01, Issue 1&lt;&gt;()  Instructions for continued Airworthiness: Doc. No. JBLS-M-0134AFM-01, Issue 1&lt;&gt;()</strong></td>
<td><strong>Prior to installation of this change/repair it must be determined that the change/repair it must be introduced no adverse effect upon the airworthiness of the product.</strong></td>
<td><strong>04/12/2019</strong></td>
<td><strong>Active</strong></td>
<td><strong>10071909</strong></td>
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<td><strong>GUENTERT + KOHMETZ GMBH</strong></td>
<td><strong>LS 3-17</strong></td>
<td><strong>10071911</strong></td>
<td><strong>A310-304</strong></td>
<td><strong>A310 ADS-B Out installation</strong></td>
<td><strong>Modification Certification Program ref. MCF-A10BD-34-001 Issue 0 dated2/01/2020Master Data List ref. MDL-S-34-423-431 issue A dated 27/11/2019Master Data List ref. MDL-S-34-423-422 issue A dated 27/11/2019Master Data List ref. MDL-S-34-423-422 issue A dated 27/11/2019Master Data List ref. MDL-S-34-423-422 issue A dated 27/11/2019Master Data List ref. MDL-S-34-423-422 issue A dated 27/11/2019</strong></td>
<td><strong>Prior to installation of this change/repair it must be determined that the change/repair it must be introduced no adverse effect upon the airworthiness of the product.</strong></td>
<td><strong>04/02/2020</strong></td>
<td><strong>Active</strong></td>
<td><strong>10071911</strong></td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10071912</td>
<td>HELLION LLC (NORWAY) AS</td>
<td>REV. 1</td>
<td>AS 332 L, AS 332 L1</td>
<td>Argos II HD FLIR installation (ref. modification AHSE-70060)Revision 1 introduces an alternative prerequisite for the embodiment of the STC.</td>
<td>The Type and OISD Certification Baseline for the original product as amended by the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: CS 29 AMD4 for the following requirements: 29.1, 29.21, 29.25, 29.45, 29.49, 29.51, 29.53, 29.55, 29.59, 29...</td>
<td>Certification Programme (CP) ref. AHSE-70091 Revision C, dated 01/November 2022 AS 332 L/L1 Flight Manual Supplement (FMS) ref. H1 No.318 Revision dated 13 September 2019 Product Change Declaration ref. AHSE-70060 Revision G, dated 29 January 2024</td>
<td>The installation of Hell- One STC ref. 1006360 or Hell-One Minor Change: GSD AHSE56152 is a pre-requisite for the embodiment of this STC.Laser pointer (laser class 3B) and laser illuminator (laser class 4) shall be disabled because they are outside of the scope of this STC and part of the EASA...</td>
<td>02/02/2024</td>
<td>Active</td>
<td>10071912</td>
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<tr>
<td>10071914</td>
<td>JET AVIATION AG</td>
<td>REV. 1</td>
<td>CL-600-2B16 (604 VARIANT) CL-600-2B16 (601-3R VARIANT) CL-600-2B16 (601-3A VARIANT)</td>
<td>WLAN Activation Demonstration of immunity to IEEE 802.11 a/b/g/n/p packet and passenger PED Transmission on Bombardier CL-600-2B16 (601-3A, 601-3R, 604 Variant), for all flight phases except for low visibility APP based on EASA OM-ES-003. The WLAN Access Point (WAP) installation is not part of this STC.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25 as of AMD4 22:789, 831, 869, 1301({}&lt;1300}&lt;1257, 1259, 1315, 1353, 1360, 1431, 1501, 1529, 1581, 1585, 1707. The...</td>
<td>- MDL JBSL-M-01244AFM-01</td>
<td>The change is applicable to Bombardier CL-600 (601-2B16 model) previously fitted with properly certified WLAN Access Point system with WLAN functionality tested and deactivated (system provisions). Limitations as stated in the AFMS (601-3R, 604 Variant) for all flight phases except for low visibility APP based on EASA OM-ES-003.</td>
<td>11/11/2019</td>
<td>Active</td>
<td>10071914</td>
</tr>
<tr>
<td>10071915</td>
<td>LEE AEROSPACES, INC</td>
<td>FAA STC STO1900W</td>
<td>DHC-8-301/311-315/401 -402 DHC-8-102/103-106-201 -202</td>
<td>Heated (LH) Side Canopy Window Assembly Replacement Installation</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the products to be replaced remain applicable to this certificate/approval without any impact on the noise database. The Certification Basis for the original product as amended by the following...</td>
<td>Master Data List No. LAI-MDL-108 Revision A dated 13 September 2019 or later FAA approved revision.</td>
<td>Prior to installation of this change it must be determined that the interrelationship between this change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft. The models DHC-8-101 and DHC-8-400 are excluded...</td>
<td>05/12/2019</td>
<td>Active</td>
<td>10071915</td>
</tr>
<tr>
<td>10071921</td>
<td>FOKKER SERVICES B.V.</td>
<td>757-300 757-200PF 757-200</td>
<td>Introduction of Dual Activated Stand-Alone GNSS Interface to ATC/Transponder on B757 to meet ADS-B Out V2 Requirements. This STC introduces the installation of a dual stand-alone Freyflight1203C GPS/ SBAS receiver as a compliant GNSS source to a B757 and added V2 capabilities on the B757, for those aircraft...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: Issue 2 of CSACNS.D AC.015, CS ACNS.D AC.025, CS ACNS.D AC.030, CS ACNS.D AC.035, CS ACNS.D AC.040, CS ACNS.D AC.040, CS AC...</td>
<td>- Compliance Report CRR-00999</td>
<td>Prior to the application of the present STC it is to have previously performed EASA STC 10068216 and Fokker Services minor mod ref CP/CRR-0049</td>
<td>09/12/2019</td>
<td>Active</td>
<td>10071921</td>
<td></td>
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<tr>
<td>10071934</td>
<td>WEATHER MODIFICATION LLC</td>
<td>FAA SA02856CH</td>
<td>C90A, C90GT, C90GTT</td>
<td>STV-23: installation of weather modification equipment for modifying weather conditions; Revision 1 correct the company name.</td>
<td>The Certification Basis (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 29 AMD4 for the following requirements: 29.1, 29.21, 29.25, 29.45, 29.49, 29.51, 29.53, 29.55, 29.59, 29...</td>
<td>To be installed in accordance with Installation drawing LH219-004-LDH_C90 revision B dated/October 16th 2019. To be operated in accordance with Aircraft Flight Manual Supplements 219-025: AFMS C-05A Revision IR (for model C90A) and 219-026:AFMS C90GT &amp; GTJ Revision IR (for models C90GT &amp; GTJ). To be main...</td>
<td>When modified with this STC the aircraft can still be operated in the normal category. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the...</td>
<td>09/12/2019</td>
<td>Active</td>
<td>10071934</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>APTOZ EHF</td>
<td>REV 2</td>
<td></td>
<td></td>
<td>D0208 GPS and Transponder Upgrade in accordance with Aptozeigengenz No. C3A431044 Revision A or later. The change covered by this STC complies with the mandatory requirement for the performance and the interoperability of the system.</td>
<td>- MDL401044 revision A or later, C3A431044-n initial revision or later (&quot;n&quot;) corresponds to the customised version related to the specific applicability, since this STC covers several types: B737-300 B737-400 B737-500 B737-600 B737-700 B737-800 B737-900 B737-900ER</td>
<td>FAA approved revision.</td>
<td>02/02/2021</td>
<td>Active</td>
<td>10071912</td>
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<tr>
<td>SAAB DEFENSE AND SECURITY USA LLC</td>
<td>FAA STC STO1254AT</td>
<td>SAAB 340B</td>
<td></td>
<td>Enlarge Class &quot;C&quot; cargo compartment of SAAB 340B by relocating cargo bulkhead forward to F.5.594.1</td>
<td>The Certification Basis for the original product remains applicable to the customised version related to the specific applicability, since this STC covers several types: B737-300 B737-400 B737-500 B737-600 B737-700 B737-800 B737-900 B737-900ER.</td>
<td>- FFV Aerotech's Master Drawing Document No. B22-2, Revision E dated 10 February 1998, or later FAA approved revision. - FFV Aerotech's Airplane Flight Manual Supplement No. B22-9, dated 30 January 1997, or later FAA approved revision.</td>
<td>09/12/2019</td>
<td>Active</td>
<td>10071939</td>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA</td>
<td>SH2917SW-D</td>
<td>AS 350 B/ AS 350 B1 AS 350 BA AS 350 D AS 350 B2 AS 350 B3</td>
<td>Installation of Electric Altitude and/or Directional Gyro instruments.</td>
<td>The Certification Basis for the original product remains applicable to the customised version related to the specific applicability, since this STC covers several types: B737-300 B737-400 B737-500 B737-600 B737-700 B737-800 B737-900 B737-900ER.</td>
<td>Prior to installation of this change/repair, the aircraft must be determined flightworthy in accordance with the specific STCs related to this change.</td>
<td>11/12/2019</td>
<td>Active</td>
<td>10071970</td>
<td></td>
</tr>
<tr>
<td>TELEDYNE UK LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A380-881 A380-842 A380-841</td>
<td>Installation of GroundLink AID+ System. The approval does not include the (T-)PEDs (installation, testing and operation). Rev. 1: Transfer...</td>
<td>The Certification Basis for the original product remains applicable to the customised version related to the specific applicability, since this STC covers several types: B737-300 B737-400 B737-500 B737-600 B737-700 B737-800 B737-900 B737-900ER.</td>
<td>Master Data List ref. MOD- MEX-037 issue 1 dated 27 November 2019/Aircraft Flight Manual Supplement ref. MOD-FMS-037 issue 1 dated 27 November 2019/Instructions for continued Airworthiness ref. MOD-ICA-037 issue 1 dated 31 July 2019/later revisions of the above listed document(s) approved/ accepted/under the...</td>
<td>26/11/2020</td>
<td>Active</td>
<td>10071971</td>
<td></td>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA</td>
<td>SH4836SW-D</td>
<td>AS 350 B/ AS 350 B1 AS 350 BA AS 350 D AS 350 B2 AS 350 B3</td>
<td>Installation of tail rotor guard.</td>
<td>The Certification Basis for the original product remains applicable to the customised version related to the specific applicability, since this STC covers several types: B737-300 B737-400 B737-500 B737-600 B737-700 B737-800 B737-900 B737-900ER.</td>
<td>Prior to installation of this change/repair, the aircraft must be determined flightworthy in accordance with the specific STCs related to this change.</td>
<td>11/12/2019</td>
<td>Active</td>
<td>10071972</td>
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<tr>
<td>ELBE FLUGZEUGWERK E GmbH</td>
<td>REV 25</td>
<td>AS31-211, AS31-231 AS31-211-231 AS30-232</td>
<td>Passenger to Freight (P2F) Conversion Rev. 1: Removal of limitations. Rev. 2: AS312P2F Frame Shell Structure improvements and optional Configurations: - RF Option A RDAefined Flight Deck (new monument, additional seat) - to allow superior entry to the Class E Main Deck CargoComp...</td>
<td>The Type and OS Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Findings (C-0341-001 Refined Flight Deck D-0307-001 Access to Class E Cargo Compartment in flight F-0307-002 Non rechargeable Lithium Bat...</td>
<td>FAA approved revision.</td>
<td>FAA approved revision.</td>
<td>05/06/2024</td>
<td>Active</td>
<td>10071994</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. Date</td>
<td>FAA STC Ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10072001</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>10/07/2004</td>
<td>FAA SH7695SW-D</td>
<td>AS 350 D</td>
<td>Installation of six heavy duty tie-down rings in the aft cabin floor.</td>
<td>The Certification Basis for the original product remains applicable to the STC, subject to the additional installation noted below.</td>
<td>EASA Drawing List 350A22-1742, Revision 0, dated 17 July 2019, indicates FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/12/2019</td>
<td>Active</td>
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<tr>
<td>10072004</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>10/07/2004</td>
<td>FAA SH4154SW-D</td>
<td>AS 350 B1/AS 350 B</td>
<td>Installation of a Steerable Landing Light installation.</td>
<td>The Certification Basis for the original product remains applicable to the STC, subject to the additional installation noted below.</td>
<td>Drawing List 350A22-1724 Revision W, dated 26 November 2013, and/or FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/12/2019</td>
<td>Active</td>
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<tr>
<td>10072006</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>10/07/2006</td>
<td>FAA SR09279RC-D</td>
<td>AS 350 B3</td>
<td>Installation of a full width or half panel replacement instrument panel.</td>
<td>The Certification Basis for the original product remains applicable to the STC, subject to the additional installation noted below.</td>
<td>Drawing List 350A22-1759 Revision 0, dated 26 June 2004 or FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/12/2019</td>
<td>Active</td>
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<tr>
<td>10072015</td>
<td>ICELANDAIR EHF</td>
<td>10/07/2015</td>
<td>767-300</td>
<td>EFB Server Connectivity via GEE SMU (Ku)</td>
<td>The Certification Basis for the original product remains applicable to the STC, subject to the additional installation noted below.</td>
<td>Icelandic Continued Airworthiness Instructions, document reference GLI-767-4620-007 7 Revision R((&gt;)=7) or later revisions of the above listed document(s) approved and accepted by the EASA system.</td>
<td>Prior or concurrent modification (s): The EFB System is to be reinstalled in accordance with EASA STC 10059804. The GEE WR System (previously known asRowan44) is to be installed per EASA STC 10059804.</td>
<td>11/12/2019</td>
<td>Active</td>
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</table>
The Certification Basis for the original product as amended by the following paragraph(s):the original product as amended by the following paragraph(s): at amendment CS-23 Amd 1: 23.5; 23.10(b)(2); 23.10(b)(3); 23.133; 23.137; 23.141; 23.147; 23.151;... - Installed in accordance with document Installation Instructions 2XL155 LEF49A(1) Issue 1 dated 9th December 2019. - Installed in accordance with document: Flight Manual Supplement 2XL L52ETF18(1), Revision 1, Dated 20th November 2019 - Maintained... - Configuration: Aircraft build standard is a Beechcraft King Air 200 with the following pre-requisite STCs: STC 10048797 Rev 2 – Installation of HC-B3TN-3G propellers with standard three bladed Hartzell... Installation Instructions - L6 & L7 2XL155 LEF49A(4) Issue 1. Insta...- Installed in accordance with document: Installation Instructions - Master 2XL152 LEF49A(1) Issue 5, Installation Instructions - Master 2XL152 LEF49A(2) Issue 2, Installation Instructions - U1 L2 L3 L9 2XL152 LEF49A(3) Issue 1, Installation Instructions - L6 & L7 2XL152 LEF49A(4) Issue 1. Insta... - Configuration: Aircraft build standard is a Beechcraft King Air 200 with Pratt & Whitney Canada PT6A-41 engines with standard three bladed Hartzell... Installation of Drop Tube - King Air 200 - Installed in accordance with document: Installation Instructions 2XL158 LEF49A(1) Issue 1 and Installation Instructions 2XL158 LEF49A(2) Issue 2. Installed in accordance with document: Flight Manual Supplement 2XL 15LE E...- Configuration: Aircraft build standard is a Beechcraft King Air 200 with Pratt & Whitney Canada PT6A-41 engines with standard three bladed Hartzell... Changes to the Adaptation of Aircraft Systems and Networks-129 Installation of Crew Rest Compartment:- The change is defined with Revision 0 of Master Data List document MGL01-2500. The aircraft shall be operated in accordance with Revision H of Airplane Flight Manual Supplement document AFMS-GL01-2500. The aircraft shall be maintained in accordance with Revision 0 of Master Data List document MGL01-2500. The aircraft shall be operated in accordance with Revision H...
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<tr>
<th>STC Holder</th>
<th>Date of Issue</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>LUFTHANSA TECHNICK AG</td>
<td>13/12/2019</td>
<td>Assurance of the original product remains unchanged and remains applicable to this certificate/approval.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Active</td>
<td>1072040</td>
</tr>
<tr>
<td>AVIONITEC AG</td>
<td>16/12/2019</td>
<td>Installation of a TAT-2200 Aircraft Communication, Addressing and Reporting System (ACARS) is replaced by a TIAS Single Blade Antenna.</td>
<td>Applicable to S/N 346423/042/PM to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Active</td>
<td>1072052</td>
</tr>
<tr>
<td>GLOBAL AEROSPACE DESIGN CORPORATION</td>
<td>21/08/2020</td>
<td>Data Link ACARS and CFDP/College installation of the Data Link with Iridium system for Aircraft Communications/Addressing and Reporting System (ACARS) and Controller PFD Data Link Communication (CPDLC) Revision 1: Fully functional installation of the Data Link with Iridium system for Aircraft...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Active</td>
<td>1072057</td>
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<tr>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td>17/12/2019</td>
<td>Cabin Reconfiguration introducing 21PAX new Premium Economy seats in the forward Economy area and a cabin divider to split the FNO economy from the Premium economy and the introduction of an additional row of seats in the All-economy section. Introduction of PRSU, cabin lights modification, new carpets and c...</td>
<td>As per MODE00289_ICA1_BEL. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Active</td>
<td>1072060</td>
</tr>
<tr>
<td>SIRIUM AEROTECH B.L.</td>
<td>21/09/2020</td>
<td>ADS-B Upgrade for EMB120/this project aims equalising with CS-ACSN 2014/045 for the approval (ADS-B use in Radar Areas. For this purpose, ADS-B parameters will be configured in transponder. New WAAS/GBS GPS installed and interfaced interchangeably with the GNSS technology system...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Active</td>
<td>1072091</td>
</tr>
<tr>
<td>LUFTHANSA TECHNICK AG</td>
<td>09/02/2022</td>
<td>The Technical O/S Certification Bases for the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Following STC have to be performed in advance or in parallel - Lufthansa STC 10069118 for the installation of Skybox-galey/lavatory on the door 4 - Lufthansa STC 10068944 for the installation of the directview camera system at door 4 - Lufthansa STC 10065111 for the exchange...</td>
<td>Active</td>
<td>1072086</td>
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<td>STC Holder</td>
<td>REv.</td>
<td>Model</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
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<td>HAJUUSUK MUYENSIUK VE SOLUTIONS</td>
<td>REV. 1</td>
<td>A320-232</td>
<td>Installation of a 180 passengers interior arrangement.Revision 1: administrative re-issuance, technical documentation and references corrected.</td>
<td>The Certification Basis for the original product as amended including additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: C5 25.817, 25.819, 25.870.</td>
<td>Master Drawing UW-M32-TS1160</td>
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<tr>
<td>ICELANDAIR EHF</td>
<td>757-200</td>
<td>Installation of 82C11Y Passenger configuration. This change introduces a cabin reconfiguration with 82C11Y passenger seating configuration and in-seat passenger systems to accommodate 757-200.</td>
<td>The Certification Basis for the original product as amended including additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: following C5 25 paragraph(s) are at amend: 1. 25.15(a)/25.011(a)/25.103(5)</td>
<td>- Top Collector Drawing ref. TC5-757-2500-527.</td>
<td>Minimum Master Equipment List Supplement, ref. MNL65-V003-25-009 Issue B, dated 31 May 2021 or later revisions of the above listed document(s).</td>
</tr>
<tr>
<td>SABENA TECHNICS BOD</td>
<td>REV. 2</td>
<td>A350-941</td>
<td>Economy seats installation. This modification installs Y/C seats in mid- and all passenger areas, together with in-seat IFE syststms, and life vest in the tamper proof seat pouch Revision 1: MSN limitaion removed Revision 2: seat replacement due to material obsolescence of backrest and kick panel.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any in...</td>
<td>- Master Data List, ref MDL-5-25-4263 Issue C, dated 08 June 2021 - Instruction for Continued Airworthiness, ref: ICA-V0003-25-009 Issue B, dated 08 June 2021 - Minimum Master Equipment List Supplement, ref: MNEL5-V003-25-0200 Issue B, dated 31 May 2021 or later revisions of the above listed document(s).</td>
</tr>
<tr>
<td>ANLAVI</td>
<td>737-900ER</td>
<td>EFB Hardware Power AID installation, installation of provision for hardware to host tablet EFB with power and data connectivity.</td>
<td>The Certification Basis for the original product as amended including additional or alternative airworthiness requirements Special Condition (s): CRI-F-31 FTChe following paragraph(s) at a later amendment: 25.29 Weight Limits 25.27 Center of gravity (25.31) &amp; (2(b)) Strength and deformation 25.307(...</td>
<td>Nawaro, Aircraft Flight Manual Supplement, doc. No. 0118-AMFS-B737 revision A, dated 04 April 2019</td>
<td>None</td>
</tr>
<tr>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>EMB-500</td>
<td>Exchange of existing Garmin GTX33D and/or GTX33i for Garmin GTX33D or ELTs and GTX33 w/ES using existing Transponder antennas</td>
<td>The Certification Basis for the original product as amended including additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: C5-22 Amendment 4.23.29, 301, 230, 307, 323, 351, 403, 407, 401, 613, 625, 627.</td>
<td>Installation, Operation and Maintenance Supplement.</td>
<td>Limited to Embraer EMB-500 aeroplanes which are factory equipped with Garmin GTX33D and G1000 Flight Deck. The Garmin G1000 Flight Deck must be configured with G1000 Load 63.7 avionics software and other higher version. Prior to installation of this change/repair it must be determined that...</td>
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<td>H4 AEROSPACE (UK) Ltd.</td>
<td>REV. 1</td>
<td>717-200</td>
<td>Honeywell ADS-B Transponder Upgrade on Boeing 717 Aircraft for Honeywell Systems &amp; Services introduces an H4A MMEL supplement as an addition to the H4 Aerospace EASA STC.</td>
<td>The Certification Basis for the original product as amended including additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: C5-22 Amendment 4.23.29, 301, 230, 307, 323, 351, 403, 407, 401, 613, 625, 627...</td>
<td>- Master Minimum Equipment List ref. H4AMMEL001, Issue 2 dated 27/10/2020</td>
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<td>10072134</td>
<td>AEROMED AEROSPACE &amp; MECHANICAL ENGINEERING SERVICES INC. WALTER STARZACHER GMBH</td>
<td>A330-343</td>
<td>A330 Monument Integration. Integration of Deltan Avionics monuments in Air Canada A330-300 aircraft.</td>
<td>- Technical Interface Document TID: M330001015_MHU_0 512018 Revision 0- Master Data List, reff: EOD_M330001015_MHU_12 02019 Revision: 0- Instructions for Continued Airworthiness (Supplement), reff: ICA_M330001015_MHU_1 2020/2019 Revision for later revisions of the above listed document(s) approved/ accepted...</td>
<td>Adaptation of the passenger cabin and installation of aircraft structural system provisions necessary for the monuments of this STC, as defined in TID_M330001015_MHU_012018 (as referenced), must be accomplished concurrently with this STC prior to release to service. Prior to installation of this change/repair any other previously installed change/repair and repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10072149</td>
<td>GULFSTREAM AEROSPACE Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBLITY</td>
<td>GV-SP GV X GV</td>
<td>Installation of Avionics T-310 KU SATCOM HSD System which supplies passenger information and entertainment services.</td>
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<td>10072155</td>
<td>AEROLITE AG</td>
<td>PC-24</td>
<td>EMS Floor KitThis Major Change to Type Design covers the introduction of the FreightFloor Inst. PIN 024-0280001-501 and/or Aircraft Freighter Inst. PIN 024-034993-501. The Aircraft Freighter installation can be installed as alternative to the freight/flight floor installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>This approval covers the floor installation for a cabin with a volumedimensional equal or not greater than the one of the TC product. Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10072159</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td>REV. 1</td>
<td>737-900ER 737-900 737-800 737-700 737-600 737-500 737-400 737-300 737-300 737-300 737-100</td>
<td>Wi-Fi WAVES installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for the introduction of the certification basis document(s) at a later amendment: EASA CS-25 Amendment 2325.301, 303, 305, 307, 321, 337, 365, 561, 601, 603, 605, 609, 611, 613, 610, 625, 771, 899, 1301, 1339, 1353, 1357, 1501, 1519...</td>
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<td>10072161</td>
<td>SPECTRUM AEROMED, LLC</td>
<td>Faa STC SA4400CH</td>
<td>PC-6/2B/H4</td>
<td>EMS conversion</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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### List of EASA Supplemental Type Certificates

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<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Related Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>CONCORDE BATTERY CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>EMB-505 (PHENOM 300)</td>
<td>Installation of Concorde valve regulated Lead-Acid batteries replacement of aircraft main batteries</td>
<td>1. Document no. 5-0678, Master Drawing List, Embraer 505 Battery/upgrade, Revision NC dated 21 March 2019, FAA approved 30 April 2019 or other FAA approved revision.2. Instructions for Continued Airworthiness (ICA). Document No. 5-0678, Revision NC dated 21 March 2019. FAA accepted 29 April 2019, or</td>
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<td>GAMA AVIATION (ENGINEERING) LIMITED</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>B300C</td>
<td>Installation of a Pressure Bulkhead Union Fitting.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA00956DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>SAAB 2000</td>
<td>Installation of Standalone ADS-B (Out) into Saab 2000 aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA00956DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged</td>
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<td>FR AVIATION Ltd.</td>
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<td>FAN JET FALCON SERIES D</td>
<td>ADS-B Out Change to installed equipment to comply with ADS-BOut requirements</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA00956DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Airworthiness and operating limits are unchanged.</td>
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<td>TEXTRON AVIATION Inc.</td>
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<td>FAA</td>
<td>B200GT</td>
<td>Installation of a Fuel Temperature Indication System on Textron Aviation, Inc.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA11228DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged.</td>
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<td>Société Air France S.A.</td>
<td>777-300ER</td>
<td>Completed Osnabrück Modification 177-17-007 Commercial cabinet configuration for AIR FRANCE Boeing 777 CO2 Freefilcting Business Class (i). Premium Economy (W) and Economy (J) classes seat replacement, rearrangement and monuments replacement. Installation of new-in-flight...</td>
<td>The Certification Basis for the original product as amended theflyswimming additional or alternative arrogance/height requirements(Special Condition (a)-CRJ D-001) Installation of inflatable restraint systems (JAR 25.862 and/AR 25.785 at Change</td>
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<td>AEROTEC &amp; CARAIBES - OCEAN INDIEN (COI) 14J Cabin Blackhawk A330 Seat refurbishment - This modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>Rocky Mountain Aircraft</td>
<td>TCCA S02-1 ISSUE 6</td>
<td>Installation of substitute passenger cabin and cargo baggage compartmentloor panels.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise level and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
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<td>Blackhawk Aerospace Technologies</td>
<td>FAA ST01853WI</td>
<td>Installation of Garmin GTS Traffic Sensor with Garmin GTX 3000 Mode STTransponders and ADS-B Out.Rev. 1 corrects the Original STC Number.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated noise level and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
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<td>J &amp; C AIRCRAFT DESIGN</td>
<td>A320-214</td>
<td>Re-activation of ISPS and associated seat change. This Change is to re-activate and adapt the In-Seat Power Supply (ISPS)System on Airbus A320-214 aircraft post modification of currently installed B/E Aerospace Pinnacle 10191610 Series passenger seats in accordance with OEM Service Bulletin (SB) Collins...</td>
<td>The Certification Basis for the original product as amended theflyswimming additional or alternative arrogance/height requirements(The following paragraph(s) at a later amendment: CS 25.851, 28.853,25.869, 25.1431, 25.1529 all amendment 22. The requirements for environmental protection and the associated...</td>
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<td>MICRO AERODYNAMICS, INC.</td>
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<td>Boeing Technical Services, UK Ltd Instructions for Continued Airworthiness (ICA) B10878-ICA000 Revision B Boeing Technical Services UK Ltd Master Data List No. B10878, MDL00 (1st) Revision of the above listed document(s) approved/accepted by EASA.</td>
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<td>The following are prerequisites for this STC Boeing Technical Services UK Ltd Minor Modification No. 2501A Boeing Master/Change No. 2500MKH105 will require EASA approval. Before this STC is installed in the affected helicopter(s) it will be added to an EU register. The following Boeing Technical Services UK Ltd documents define the system.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the original product.</td>
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<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertificatednoise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>- Master Data List A330 Air Lines - Ref. MDL-9-00-4188 Issue B, dated 15 January 2020: Revision for Continued Airworthiness, FAA-CAFAF-00000-004 (Issued dated 15 January 2020). 1st Revision of the above listed document(s) approved/accepted by EASA.</td>
<td>Boeing Technical Services, UK Ltd Instructions for Continued Airworthiness (ICA) B10878-ICA000 Revision B Boeing Technical Services UK Ltd Master Data List No. B10878, MDL00 (1st) Revision of the above listed document(s) approved/accepted by EASA.</td>
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<tr>
<td>MEDITERRANEAN AVIATION COMPANY LTD. (MEDAVIA)</td>
<td>10072257</td>
<td>REV. 1</td>
<td>737-700</td>
<td>New Maximum Take-Off WeightReduction of the certified maximum take-off weight of the aircraft to 56,472 kg, and maximum landing weight to 53,388 kg. Rev. 1: Added aircraft configuration with CFM56-7B2E2 engines.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection are unchanged but the threshold/weight approval involves a change with an impact on the noise database and/or emission levels.</td>
<td>Airplane Flight Manual Supplement MODEO00292_AFMS1 rev 2(&lt;&gt;) or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>- Applicable to Boeing 737-700 airplanes, equipped with engines and/CFM56-7B2E2 or CFM56-7B2E2 engines (including interim engine configuration). - Aeroplane must be operated as per AFMS MODEO00292, AFMS1 rev 2 or later. Prior to installation of this change/repair it must be determined that the interference...</td>
<td>20/11/2020</td>
<td>Active</td>
<td>10072257</td>
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<tr>
<td>COMTECH GLOBAL AVIATION SERVICES S.</td>
<td>10072259</td>
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<td></td>
<td>Introduction of FAX, E/C Seat Cushions. This modification introduces new cushions to the standard economy class(E/C) seats keeping existing dress cover fabric materials.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>CTG-MS5040-777 MODIFICATION SUMMARY:&lt;&gt; or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Comtech bottom seat cushions P/N: 864984-401CTG are not approved for installation in front row seats. Original seat manufacturer installation instructions remain applicable after installing the Comtech bottom seat cushion P/N: 864984-401CTG.</td>
<td>23/01/2020</td>
<td>Active</td>
<td>10072259</td>
</tr>
<tr>
<td>SAFRAN CABIN INC.</td>
<td>10072281</td>
<td></td>
<td>CL-600-2B19</td>
<td>Installation of Reinforced Cockpit Doors with Flight Deck Access System. Rev. 2. Introducing MMEL. Original STC Number: FAA STC ST01416LA.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>C &amp; D Interiors Controlled Drawing List, 4631000, Revision B, dated 03 December 2002, or later FAA approved revision. Safran, IMEL Supplement to CL-600-2B10-2C10-2D15/-2D4/2E25 with EASA STC 1006831. Rev 1 Installed, dated 09 January 2003.</td>
<td>Equipment Functional Check Intervals as defined in FAA approved C &amp; D Interiors Report CDR4635-69, Revision A, dated November 8, 2002, “Certification Maintenance Requirements”, or any later FAA approval revision, shall not be extended. Later revisions to C &amp; D Interiors Report CDR4635-69 may be used only...</td>
<td>25/01/2020</td>
<td>Active</td>
<td>10072281</td>
</tr>
<tr>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>10072283</td>
<td>FAA SR11243DS</td>
<td>505</td>
<td>Installation of Expanded Avionics Shelf.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments indicated on FAA STC. There are requirements for environmental protection and the associated certified noise and/or emissions levels of the original product...</td>
<td>1. Master Drawing List, Report No. AA-1674. Revision A, dated November 2009, or later FAA approved revision. 1. Master Drawing List, Report No. AA-1674. Revision A, dated November 2009, or later FAA approved revision. 1. Master Drawing List, Report No. AA-1674. Revision A, dated November 2009, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/02/2020</td>
<td>Active</td>
<td>10072283</td>
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<td>STC Holder</td>
<td>STC STC ref.</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>AEROSPACE AIRWORTHINESS</td>
<td>10072284</td>
<td>03/07/2024</td>
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<td>FOKKER Active AVINCIS</td>
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<td>AINOCIS AVIATION ITALIA S.P.A</td>
<td>10072285</td>
<td>REV. 2</td>
<td></td>
<td></td>
<td>CL-415 ADS-B OUT CAPABILITY Introduction of the Broadcast Dependent Surveillance – Broadcast Out depravity (ADS-B Out) [Rev. 1]</td>
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<td>Active</td>
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<tr>
<td>FOKKER SERVICES B.V.</td>
<td>10072279</td>
<td>REV. 1</td>
<td>737-400, 737-300, 737-200, 737-400</td>
<td></td>
<td>B737 - Antenna Structural Provisions</td>
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<td>Active</td>
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<tr>
<td>3S CERTIFICATION, LLC</td>
<td>10072283</td>
<td>B200</td>
<td>FAA STC 8A18820WI</td>
<td></td>
<td>Bendeflying Aero/Vue Avionics Systems Upgrade - B200</td>
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<td>STC Holder</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>APTO EHF</td>
<td>737-700</td>
<td>Indum-Antenna Installation - Structural installation of Barrier System/Indum antenna Part Number 567-1575-365.</td>
<td>- Instructions for Continued Airworthiness ICA231058-1 Revision A - Change 1 repair approval, CRA231058 Revision Aor later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>- This STC covers only the structural installation of the antennas Functional and EMC aspects and equipment qualification are not covered by this STC - Inspections must be accomplished in accordance with the instructions ICA231058-1 Revision A, or later. Prior to installation of this change/repair it...</td>
<td>28/11/2020</td>
<td>Active</td>
<td>10072296</td>
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<tr>
<td>B/E AEROSPACE, Inc.</td>
<td>FAA ST0419AT</td>
<td>Installation of the Rockwell Collins Automatic Dependent Surveillance - Broadcast (ADS-B) Out System.</td>
<td>1. Master Drawing List (MDL) 2-190235-001, Rev. A, dated 8 December 2017, or later FAA approved rev.2.</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this design change and any other previsously installed design change and/or repair will introduce aero dynamic effect upon the airworthiness of the product.</td>
<td>10/02/2020</td>
<td>Active</td>
<td>10072295</td>
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<td>PMV ENGINEERING</td>
<td>A310-314/315</td>
<td>Certification Basis for the original product as amended/ revised 2019.</td>
<td>Certification Plan ref. AVI-01830-CIR-R02 dated 28 Oct. 2019</td>
<td>This approval has considered the configuration described in the Certification Plan ref. AVI-01830-CIR-R02 dated 28 Oct. 2019;</td>
<td>28/01/2020</td>
<td>Active</td>
<td>10072113</td>
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<tr>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td>REV. 1</td>
<td>New Maximum Takeoff Weight/Reduction of the certified maximum take-off weight of the aircraft 737-700 1999 kg/-1: 1. Addied a co configuration with CMT56-782E engine.</td>
<td>- Applicable to Boeing 737-700 airplanes equipped with winglets and CFA56-782E / CFE56-792E engines (including interim engineconfiguration). - Aircraft must be operated as parAFMS MODE2091, AFMS rev 3 or later. Prior to installation of this change/repair it must be determined the airworthiness of the product.</td>
<td>20/11/2020</td>
<td>Active</td>
<td>10072230</td>
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<td>Q.C.M. DESIGN GmbH</td>
<td>FALCON 7X</td>
<td>Removal of A/C Cabin &amp; Lav Ashtrays . Smoking is not permitted on the aircraft.</td>
<td>- AFM DF8X-QCM25-001, M1-ICA231058-1 M1-ICA231058-1 Revision Aor later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined the airworthiness of the product.</td>
<td>30/01/2020</td>
<td>Active</td>
<td>10072321</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10072325</td>
<td>MAP AIRCRAFT PART 21 A8</td>
<td>737-800</td>
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<td>Stanchem Litter System &amp; Oxygen Module Installation</td>
<td>The Certification Basis for the original product as amended by this containing: additional or alternative airworthiness requirements (special condition(s)): D-011/05/03/13 Medical Evacuation. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product.</td>
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<td>31/01/2020</td>
<td>Active</td>
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<tr>
<td>10072356</td>
<td>UNICAL AERO INC</td>
<td>Faa STO72010A</td>
<td>A321-272NX A321-252NX-272NX-25NX A321-252NX-253NX-271NX A321-231-231-231-251-252 A321-111-112-131-211-212 A320-271ND-272N-273N A320-231-231-231-251-251-232 A320-2101-231-212-214-216-232 A320-121-121-121-121-121 A320-121-121-121-121-121 A320-121-121-121-121-121 A320-121-121-121-121-121 A320-121-121-121-121-121</td>
<td>Installation of Unical Aero USB-PED Charging System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>10072329</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>CAA UK RESPONSIBILIT</td>
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<td>31/01/2020</td>
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<tr>
<td>10072331</td>
<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>REV. 1</td>
<td>A340-600</td>
<td>A340-600 LOPA Installation Rev 1: Transfer of STC from GVH Aerospace Ltd - United Kingdom, to GVH Aerospace (Ireland) Limited</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10072343</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>Faa STC SR09511D3</td>
<td>EC 130 T2 EC 130 B4 AS 350 B3 AS 350 B2</td>
<td>Installation of LED Taxi and Landing Lights with Optional Pulse</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC (SR09511D3). The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>For AS350B2 and B3 models: - Instructions for Continued Airworthiness 350MS-044 Rev. A dated 12 December 2019, or later ODAs or AEG accepted revision. - Rotorcraft Flight Training Manual Supplement 350MS-550 Rev. IR dated 12 December 2019, or later ODAs approved revision. For EC 130T2 and B4 models: - Instructions.</td>
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<td>STC Holder</td>
<td>FAA STC No.</td>
<td>Model</td>
<td>Modification</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>STC Holder</td>
<td>FAA STC No.</td>
<td>Model</td>
<td>Modification</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AVIATION COMMUNICATION</td>
<td>FAA STC0251A</td>
<td>737-800-700-800-900-9000</td>
<td>Modification to the existing Mode S Transponder system and option replacement of ATC/CA RevControl Panel to add NEXTGENADS B Out</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval, except where amended by additional or later amendments</td>
<td>1. Aviation Communication &amp; Surveillance Systems Index of Master Data, Drawing No. 8516377, Revision - dated 2 November 2016, or later FAA approved revision. 2. Instructions for Continued Airworthiness (ICA) and Airplane Flight Manual Supplement (AFMS) are called out in the corresponding FAA.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2020</td>
<td>Active</td>
<td>10072345</td>
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<tr>
<td>NORTWEST AEROSPACE</td>
<td>FAA SA0756D5</td>
<td>100, A100, A100A, B100</td>
<td>Installation of an S-TEC 01326 Digital Flight Control System (DFCS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Master Data List No. ST-964-ML-0001, Revision A, dated 12.11.2016/Flight Manual Supplement, ST-964-AMS-0001, Revision - dated 06.03.2019.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/02/2020</td>
<td>Active</td>
<td>10072346</td>
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</tr>
<tr>
<td>S-TEC CORPORATION</td>
<td>FAA SA0756D5</td>
<td>100, A100, A100A, B100</td>
<td>Installation of an S-TEC 01326 Digital Flight Control System (DFCS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Master Data List No. ST-964-ML-0001, Revision A, dated 04.09.2019/Flight Manual Supplement, ST-964-AMS-0001, Revision - dated 21.08.2019.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2020</td>
<td>Active</td>
<td>10072379</td>
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<tr>
<td>S-TEC CORPORATION REV. 1</td>
<td>FAA SA0756D5</td>
<td>100, A100, A100A, B100</td>
<td>Installation of an S-TEC 01326 Digital Flight Control System (DFCS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Master Data List No. ST-964-ML-0001, Revision A, dated 04.09.2019/Flight Manual Supplement, ST-964-AMS-0001, Revision - dated 21.08.2019.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2020</td>
<td>Active</td>
<td>10072379</td>
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<tr>
<td>SCANDINAVIAN AVIONICS</td>
<td>DORNIER 328-300</td>
<td>04/02/2020</td>
<td>Upgrade of existing dual MODE-S Transponder system with ADS-B functionality in order to comply with operational requirements</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged.</td>
<td>Aircraft Flight Manual, Doc. Ref: 2118360 AM, Issue: 1 Continued Airworthiness, Doc. Ref: 2118360 CA, Issue: 1.2, later revisions of the above listed document(s) approved/accredited under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2020</td>
<td>Active</td>
<td>10072466</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>A330-203</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact...</td>
<td>- Supplemental Type Design Document: STC25-192HM Issue 1 Rev 0 dated 06 Feb 2020; Instruction for Continued Airworthiness: A-00-2010-ICA Issue 1 Rev 1 dated 27 Jan 2020; Compliance Checklist: CCLUS-192HM Rev 2 dated 06 Feb 2020 or later revisions of the above listed document(s) approved/accepted/</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2020</td>
<td>Active</td>
<td>10072459</td>
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<tr>
<td>STANDARD AERO BUSINESS AVIATION SERVICES ES, L.L.C.</td>
<td>BD-700-1A11 BD-700-1A10</td>
<td>Installation of a Honeywell JetWave Satcom System with associated Ka-Band Radome</td>
<td>Approved StandardAero Airplane Flight Manual Supplement No: 1033377 Revision (C) dated 27 November 2019; StandardAero Instructions for Continued Airworthiness No: 1033376 Revision (C) Airworthiness Limitations dated 11 June 2019; Accepted: StandardAero Master Data List No: 1033284, Revision (D) dated 10 Dec.</td>
<td>This STC requires prior embodiment of a T-PED STC, Service Bulletin reissue of EASA STC no: 10069212. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherwise previously installed change and/or repair will introduce no adverse effect.</td>
<td>14/02/2020</td>
<td>Active</td>
<td>10072539</td>
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<tr>
<td>ASI AVIATION</td>
<td>B300</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact...</td>
<td>Master Data List ref. DL n° 047 Rev. 02 dated 29/01/2020 Operation in accordance with Aircraft Flight Manual Supplement Ref: PCH-350-95000 Rev. 02 dated 29/01/2020 Maintain in accordance with Beechcraft Corporation approved ICA product Ref: 130-59031-539, Original date: 01.05.2015 for increase...</td>
<td>Changes to the individual aircraft Airworthiness Limitations will need to be EASA Approved Aircraft with a maximum MTOW of 17,500 lb are eligible only for a Restricted Category Certificate of Airworthiness.</td>
<td>17/02/2020</td>
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<td>10072562</td>
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<td>STC Holder</td>
<td>Active/Inactive</td>
<td>System</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>10072564</td>
<td>Active</td>
<td>STC007</td>
<td>Installation of Butterfield Industries Oil</td>
<td>The Certification Basis (CB) for</td>
<td>1. Installation Date Package BI 8020/(v)+</td>
<td>Prior to installation of</td>
<td>19/02/2020</td>
<td>Active</td>
<td>10072564</td>
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<td>applicable to this certificate/</td>
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<td>Installation (Major Change) and its</td>
<td>MM-2157-00000, Issue 4, dated</td>
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<td>alternative airworthiness</td>
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<td>between this change/repair and</td>
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<td>FAA STC</td>
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List of EASA Supplemental Type Certificates

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<th>STC Holder/Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>MT-PROPELLER ASW 27 – installation of a turbojet</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any im...</td>
<td>1. Collins Certification Index as listed on AML No. ST01459WI-D, revised 26 August 2014, or later FAA approved amendment.</td>
<td>Prior installation and approval of a Collins MT-PROPELLER ASW 27 STC is valid for aircraft types and models approved by EASA an...</td>
<td>08/02/2020</td>
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<td>ARMSTRONG AEROSPACE Inc.</td>
<td>A321-217N/218N/219N/220N/221N/222N</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any im...</td>
<td>1. Armstrong Aerospace Master Data List. Doc. No. DC104-8098-00, Revision C, dated 29 May 2019, or later FAA approved ...</td>
<td>Modification in accordance with this STC requires that the aircraft be...</td>
<td>26/02/2020</td>
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<td>EICHELSDORFER GmbH FLUGZEUGBAU</td>
<td>ASW 27 – installation of a turbojet</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Finding(s): 22.975, 22.977. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any im...</td>
<td>Operation in accordance with Aircraft Flight Manual Supplement, Eichelsdörfer Flugzeugbau document ASW 27 STC PIR FL, annexation 4 dated September 09, 2018, Airworthiness Limitations as per Instruction for Continuing Airworthiness Supplement, Chapter 4, Eichelsdörfer Flugzeugbau document ASW 27 STC PIRs.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelation...</td>
<td>26/02/2020</td>
<td>Active</td>
<td>10072596</td>
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<td>FOKKER SERVICES B.V.</td>
<td>ADS-B Out on the Boeing 737 Classic</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any im...</td>
<td>The change is defined through issue 4 of MDL document 8737-ADS-B-OUT-MDL-0002. The aircraft shall be operated in accordance with issue 1 of Airplane Flight Manual Supplement document 8737-ADS-B-IFM-S-G02 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the...</td>
<td>27/02/2020</td>
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<td>10072595</td>
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An agency of the European Union
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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>10072642</td>
<td>SIRIUM AEROTECH S.L.</td>
<td>A310-325</td>
<td>A310-324</td>
<td>New Supplement to AFM for ADS-B Upgrade in Airbus A300/310 Airplanes</td>
<td>The Certification Basis for the original product as amended by following additional or alternative airworthiness requirements Section 4 of CS-ACNS Subpart D Issue 2 for ADS-B OUT Approvals. The requirements for environmental protection and the associated certified noise and/or emissions levels of the...</td>
<td>Limitations as per: AFM Supplement SA19-695-AMMS Rev00 Prior to installation of the change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/03/2020</td>
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<td>10072643</td>
<td>SAA SOLUTIONS FOR AVIATION, S.L.</td>
<td>S50</td>
<td>S500</td>
<td>ADS-B OUT INSTALLATION (according to CS ACNS Issue 2)</td>
<td>Certification basis for this design change are the original certification basis, as defined by the type certificate(s) and ammended by the following CS25 Amendment 23 paragraphs: 25.1, 25.21, 25.23, 25.25, 25.27, 25.29, 25.301, 25.303, 25.305, 25.307, 25.321, 25.337, 25.361, 25.371, 25.381, 25.401, 25....</td>
<td>Modification in accordance with Master Document List, 50010-RPT02 Ed 20 Operation in accordance with Flight Manual Supplement, 50010-HD001 Ed 2 Instructions for continued airworthiness in accordance with Maintenance Manual Supplement, 50010-HD002 Ed 1 for later revisions of the above listed document(s).</td>
<td>Prior to installation of this change/repair it must be determined that the boundary relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/03/2020</td>
<td>Active</td>
<td>10072643</td>
</tr>
<tr>
<td>10072649</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>C90A, C90GT, C90GTI</td>
<td>Installation of either single or dual Garmin G8AS antennas.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any...</td>
<td>Installation in accordance with Avionik Straubing GmbH Instructions for Installation Doc. No: ASR-2019-074-PI-10-00, Revision 00, date 19/11/19. Instructions for Continued Airworthiness in accordance with Avionik Straubing GmbH Doc No: ASR-2019-074-ICA-11-00 Revision 00, date 31/01/2020 or later revisions...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/03/2020</td>
<td>Active</td>
<td>10072649</td>
<td></td>
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<tr>
<td>STC Holder</td>
<td>Revision Level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>BOMBARDIER Inc.</td>
<td>REV. 1</td>
<td>BD-700-2412</td>
<td></td>
<td>Installation of Emergency Vision Assurance System (EVAS) Rev 1: Update of the STC holder address.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>Modification Data Summary, Bombardier Inc. document No: RAZ-BM700-16B.Revision NC dated 29 November 2019 Instructions for Continued Airworthiness, Bombardier Inc. document No: EVS256880-ICA, Revision 2 dated 22 November 2019.Aircraft Flight Manual Supplement, Bombardier Inc. document No: ICA-290002 Rev. 1</td>
<td>Prior to installation of this change/repair it must be determined if the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/04/2023</td>
<td>Active</td>
<td>10707032</td>
</tr>
<tr>
<td>TUI AIRLINES BELGIUM N.V.</td>
<td></td>
<td>767-300</td>
<td></td>
<td>Installation of a wireless IFE system on B767-300 (Change Number: JAF-STC-0003-0001 Rev. 1).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraphs at a later amendment C5-25 at alt: 12 ft C5-26 25.301(a)(ii): 25.303, 25.305(a)(ii), 25.310(a), 25.321, 25.341, 25.364(b)(iii), 41(g), 41(h), 25.471 (a) ...</td>
<td>- Certification Plan ref. JAF-DOA-CP-44M-0004 R07, dated 25/11/2019 - AFM Supplement ref. AFMS-JAF-DOA-44M-0004 R06, dated 16/09/2019 - ICA Report JAF-DOA-44M-0004 R06, dated 10/12/2019 - MDL ref. JAF-STC-0003-0011 Rev. 000 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined if the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/03/2020</td>
<td>Active</td>
<td>10707255</td>
</tr>
<tr>
<td>SEBALD LUFTFAHRTTECHNIK NISCHER BETRIEB(LTB)</td>
<td></td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td></td>
<td>TM S04 Dynamic balancing/ Dynamic balancing of a retractable propulsion system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>TM-S04 Auswechslung von Klapptriebwerken, Ausgabe 02.07.2019 (&lt;, &gt;) or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>None</td>
<td>04/03/2020</td>
<td>Active</td>
<td>10707283</td>
</tr>
<tr>
<td>BABCOCK MISSION CRITICAL SERVICES DESIGN AND COMPLETIONS LIMITED</td>
<td></td>
<td>AB139, AW139</td>
<td></td>
<td>Martin Baker Rotating Seat Installation: This STC installs a Martin Baker Rotating and Traversing Crashworthy Seat (PN MBCS6930) into the cabin of the AB/AW139 helicopter models.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Certification Plan ref. JAF-DOA-CP-44M-0004 R07, dated 25/11/2019 - AFM Supplement ref. AFMS-JAF-DOA-44M-0004 R06, dated 16/09/2019 - ICA Report JAF-DOA-44M-0004 R06, dated 10/12/2019 - MDL ref. JAF-STC-0003-0011 Rev. 000 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The Leonardo Helicopters Removable Floor Panel PN: 3P5332A00112 is a prerequisite for the installation of this STC. Prior to installation of this change/repair it must be determined if the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>04/03/2020</td>
<td>Active</td>
<td>10707266</td>
</tr>
<tr>
<td>AVIATECH &amp; CONSULTING, SOCIEDAD LIMITADA</td>
<td>REV. 1</td>
<td>S2R-T660</td>
<td></td>
<td>Installation of Computerized Fireproof Installation of computerized firegates, used and approved already on TRAFO TRACTOR 502 and 802 Series under STC EASA 10054617, with an adder for the tractor of the S2R-T660 Rev. 1: Transfer from TRABAOS NEREOS ESPEJO S.L. to AVIATECH &amp; CONSULTING SOCIEDAD LIMITADA</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>1) Master Drawing List, MDL-01-18 Rev. 0, Fecha: 30/09/2019 2) Boletin de Ingenieria - Instrucciones de Instalacion y conservabilidad Continuida B6/1-18 Rev. 1, Fecha: 30/06/2019 3) Suplemento al Manual de Vuelo SM/01-18 R1, Fecha: 30/06/2019 or later revisions of the above listed document(s) approved...</td>
<td>Prior to installation of this change/repair it must be determined if the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2022</td>
<td>Active</td>
<td>10707276</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10072690</td>
<td>AEROMECH Inc.</td>
<td>REV 1</td>
<td>Installation of Garmin G950 Integrated Autopilot and Display SystemRev. 1 corrects a typo in AFM supplement reference.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: FAR 23 amdt. 34, 23.1, 23.207(b), 23.201, 23.301, 23.303, 23.305, 23.307, 23.333, 23.337, 23.341, 23.865(a)(k)(i), 23.5...</td>
<td>Installation in accordance with Master Document List (MDL) document AM0203-0200 Revision G dated 8 June 2021. Maintenance in accordance with Instructions for Continued Airworthiness (CA), document AM2000-001-00 Revision G, dated 20 December 2017. Operations in accordance with Airplane Flight Manual.</td>
<td>Prior installation of this change/repair must be determined in the EASA system.</td>
<td>24/03/2023</td>
<td>Active</td>
<td>10072690</td>
<td></td>
</tr>
<tr>
<td>10072708</td>
<td>DIAMOND AIRCRAFT INDUSTRIES, Inc.</td>
<td>TCCA</td>
<td>Installation of MTV-12-B/183-59b Propeller</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/approval are unchanged but this certification/approval involves an impact on the noise database and/or emissions levels.</td>
<td>FAA STC SA03281AT must be installed with this STC. Prior to installation of this change/repair it must be determined that the interference relationship between this change/repair and any other previously installed change and/or repair will introduce adverse effect upon the airworthiness of the product.</td>
<td>09/03/2020</td>
<td>Active</td>
<td>10072708</td>
<td></td>
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<tr>
<td>10072732</td>
<td>AAR AIRCRAFT SERVICES</td>
<td>FAA STC</td>
<td>Installation of Center Overhead Life Raft Stowage Compartments</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/approval are unchanged but this certification/approval involves an impact on the noise database and/or emissions levels.</td>
<td>AAR Aircraft Services Master Data List 1001825002, RevisionD, dated 25 April 2012.</td>
<td>Prior to installation of this design change it must be determined that the interference relationship between this design change and any other previously installed design change and/or repair will introduce noise/reffect upon the airworthiness of the product.</td>
<td>16/02/2020</td>
<td>Active</td>
<td>10072732</td>
<td></td>
</tr>
<tr>
<td>10072739</td>
<td>COMTECH GLOBAL AVIATION SERVICES B.</td>
<td>A330-342,A330-343</td>
<td>Introduction of PAX E/C Seat Cushion/This modification replaces the bottom cushions of Weber Seats Model 575series with bottom cushion P/N 864984-401CTG, maintaining the existing cover fabric materials.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/approval are unchanged but this certification/approval involves an impact on the noise database and/or emissions levels.</td>
<td>CTG-MSO545-320 MODIFICATION SUMMARY Initial Release 210202020X later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>Comtech bottom seat cushions P/N 864984-401CTG are not approved for installation in front row seats. Original seat manufacturer installation instructions remain applicable after installing the Comtech bottom seatcushion P/N 864984-401CTG.</td>
<td>12/03/2020</td>
<td>Active</td>
<td>10072739</td>
<td></td>
</tr>
<tr>
<td>10072740</td>
<td>COMTECH GLOBAL AVIATION SERVICES B.</td>
<td>A380-881</td>
<td>Introduction of PAX E/C Seat Cushion/This modification replaces the bottom cushions of Weber Seats Model 575series with bottom cushion P/N 864984-401CTG, maintaining the existing cover fabric materials.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/approval are unchanged but this certification/approval involves an impact on the noise database and/or emissions levels.</td>
<td>CTG-MSO546-380 MODIFICATION SUMMARY Initial Release 210202020X later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>Comtech bottom seat cushions P/N 864984-401CTG are not approved for installation in front row seats. Original seat manufacturer installation instructions remain applicable after installing the Comtech bottom seatcushion P/N 864984-401CTG.</td>
<td>12/03/2020</td>
<td>Active</td>
<td>10072740</td>
<td></td>
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<tr>
<td>10072752</td>
<td>SMART ELCOTTERI S.P.A</td>
<td>REV. 2</td>
<td>PA-31T (CHEYENNE/ CHEYENNE II) KIT CDA-04 for PA-31T SN 31T-7620005Revision 2: Transfer of STC to SMART Elcotteri</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate/approval are unchanged but this certification/approval involves an impact on the noise database and/or emissions levels.</td>
<td>- Master Drawing List Kit Doc. PO0500-MDL-01-00 dated 23/04/2019. Master Document List Doc. PO0500-MDL-01-00 dated 23/04/2019 Master Document List Doc. PO0500-001-00 AMD Doc. PO0500-001-00 AMD dated 25/06/2019. AFMS Doc. PO0500-001-00 AMD dated 12/02/2020. Later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>Applicable to PA-31T SN 31T-7620005 only. Prior to installation of this change/repair it must be determined that the interference relationship between this change/repair and any other previously installed change and/or repair will introduce adverse effect upon the airworthiness of the product.</td>
<td>02/07/2020</td>
<td>Active</td>
<td>10072752</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10072765</td>
<td>AERODISA S.L.</td>
<td>2</td>
<td></td>
<td>B300</td>
<td>AFM SBAS procedures for the B300 series.</td>
<td>-Master Data List AED2016-40-MDL Rev, AO AFM Supplement AED2016-40-AMFS Rev Alter later revisions of the above listed document(s) approved/accepted/published with the EASA system.</td>
<td>Limitations as per - This EASA Design Approval must be limited to the Airplane Model B300(1-4) in MSN FL-259, Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>12/03/2020</td>
<td>Active</td>
<td>10072765</td>
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<tr>
<td>10072771</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 2</td>
<td></td>
<td>680A (LATITUDE)</td>
<td>Approval of UNIFS 3000 System Installation in accordance with Scandinavian Avionics Modification 2110369 Revision 2 of this STC includes VHF COMM Extended Frequency Range. Configuration enablement of extended frequency range on VHF COMM systems, equipment prior installed.</td>
<td>Modification in accordance with Scandinavian Avionics Engineering Order No. 2110369-ETO Iss. 2, dated 12.11.2020 and Master Document List No. 2110369 DOC Iss. 2 dated 12.11.2020. Operate in accordance with Scandinavian Avionics Flight Manual Supplement ref 2110369-AM (i), as defined in Master document 2.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>24/11/2020</td>
<td>Active</td>
<td>10072771</td>
</tr>
<tr>
<td>10072772</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>680A (LATITUDE)</td>
<td>Approval of Aircraft Audio System Modification in accordance with Scandinavian Avionics Modification 2111323.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 2, except following paragraphs: 25-29, 301, 303, 305(a)(b)(c)(d), 307, 321, 335, 337, 341, 365, 371, 561, 581. Installation in accordance with Scandinavian Avionics Engineering Order No. 2113232-ETO Iss. 1, dated 06.03.2020 and Master Document List No. 2113232 DOC Iss. 1 dated 06.03.2020. Operate in accordance with Scandinavian Avionics Flight Manual Supplement ref 2113232-AM Iss 2, dated 11.12.2019</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>13/03/2020</td>
<td>Active</td>
<td>10072772</td>
<td></td>
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<tr>
<td>10072820</td>
<td>EMBRAER ENGINEERING &amp; TECHNOLOGY US, INC</td>
<td>FAA STC SA4471AT</td>
<td>EMB-500</td>
<td>Installation of a Tire Pressure Sensing Kit</td>
<td>- Master Data List (M DL) 17007-001, Rev. 0, dated 21 November 2018 - Instructions for Continued Airworthiness (ICA), Document CA/17007-030, Rev. 0, dated 21 November 2018.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>16/03/2020</td>
<td>Active</td>
<td>10072820</td>
<td></td>
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<tr>
<td>10072821</td>
<td>IKHANA GROUP, LLC</td>
<td>FAA STC SA0256LA</td>
<td>DHC-6 SERIES 400-500 DHC-6 SERIES 320 DHC-6 SERIES 310 DHC-6 SERIES 300 DHC-6 SERIES 210 DHC-6 SERIES 200 DHC-6 SERIES 110 DHC-6 SERIES 100 DHC-6-100-200-300-400</td>
<td>Installation of Passenger Utility Seats Rev. 1: Name Change from &quot;Mhahana Group Inc.&quot; to &quot;Mhahana Group LLC.&quot;</td>
<td>- IKHANA Aircraft Services Utility Seat Installation Drawing No. SU00-0011, Revision A, dated 2 December 2015 - Design Approval of Passenger Utility Seats per IKHANA Aircraft Services Master Drawing List, Document No. ER354-1000, Revision A dated 3 December 2018.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>09/06/2021</td>
<td>Active</td>
<td>10072821</td>
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<tr>
<td>10072822</td>
<td>AVIONIC STRAUBING ENTWOERING Gmbh</td>
<td>FAA STC SA04471AT</td>
<td>EMB-500</td>
<td>ADS-B approval. Exchange of existing single unit ACSS RCZ-852 transponder/dua</td>
<td>Installation, Operation and Maintenance according to Avionic Stealing Master Document No. ASR-2019-073-MDL-01-00, revision 00, dated 27 February 2020 or later revisions of the above listed document(s) approved/accepted/published with the EASA system</td>
<td>Limited to Embraer EMB-500 ailements which are factory equipped with single unit ACSS RCZ-852 and a Garmin G1000 or Garmin G3X900 Flight Deck. The Garmin G1000 Flight Deck must be configured with G1000 Load 85.20.a or higher version. The Garmin G3X900 Flight Deck must be configured with G3X900 Load 13.25a.</td>
<td>24/03/2020</td>
<td>Active</td>
<td>10072822</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10072824</td>
<td>SIRIUS AEROTECH S.L.</td>
<td>5500</td>
<td>560X, 560L, 560, 550</td>
<td>New Supplement to the EASA AFM for RNAV 1 Approval</td>
<td>The Certification Basis for the original product as amended by additional or alternative airworthiness requirements. The requirements for environmental protection and the associated certified noise and/or emissions levels of...</td>
<td>- Master Data List SA19-835-MDL Rev 001 approval</td>
<td>FAA STC 10072824</td>
<td>Limitations as per AFM Supplement SA19-835-AMF Rev 001. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/03/2022</td>
<td>Active</td>
</tr>
<tr>
<td>10072827</td>
<td>E A D AEROSPACEMANAGEMENT</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>KA-BAND SATCOM Skirt - Instructions for Continued Airworthiness</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>- Certification Program ref. 01599-EAS-A-CFM-01, Rev B (13 March 2020)</td>
<td>FAA STC 10072827</td>
<td></td>
<td>The ICA generated in the frame of this approval supersede the ones generated for STCs 10067720 Rev 1 (on B747), 10062160Rev 3 (on B777), 10063487 Rev 1 (on A340), and 10066550 Rev 2 (on A330). Prior to installation of this change/repair it must be determined that the relationship between...</td>
<td>17/03/2020</td>
<td>Active</td>
</tr>
<tr>
<td>10072830</td>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SA09757DS</td>
<td>Installation of an S-TEC 01336 Digital Flight Control System (DFCS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended or...</td>
<td>FAA STC 10072830</td>
<td></td>
<td>Prior to installation of this design change it must be determined that the relationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/03/2020</td>
<td>Active</td>
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<tr>
<td>10072831</td>
<td>ALAMO ENGINEERING GmbH</td>
<td>REV. 2</td>
<td>EMB-136BJ</td>
<td>PM-CPOLC with MK2 + CMU and VDR Upgraded Modification of FAN1861A (wrapped aircraft to add PM-CPOLC (ATNB1))</td>
<td>The Certification Basis for the original product as amended by additional or alternative airworthiness requirements. Special Condition (s): CRI-R31 Flight Recorders and Data Link Recording Issue 3 - CS-ACNS. Issue 2. The requirements for environmental protection and the associated...</td>
<td>Master Data List No. ST-976-MDL-0001, Revision A, dated 22 January 2020 Flight Manual Supplement, ST-976- AFMS-0001, Revision - dated 23 January 2020 Instructions for Continued Airworthiness, ST-976- ICA-0001, Revision - dated 21 January 2020.</td>
<td></td>
<td>Master Data List ML1682001, Issue 6 A340; A330 Approval DC-14M168Z0002 is a required element for this STC. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the...</td>
<td>18/12/2023</td>
<td>Active</td>
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<tr>
<td>10072832</td>
<td>ACC COLUMBIA Jet Service GmbH</td>
<td>REV. 3</td>
<td>690C</td>
<td>RC 690C Camera Lift Installation Rev. 0</td>
<td>The Type and O.M. Certification Base for the original product as amended by the following additional or alternative airworthiness requirements following paragraph(s) at a...</td>
<td>Master Data List ML1682001, Issue 6 A340; A330</td>
<td></td>
<td>Aircraft build standard is Twin Commander RC690 C as described in PAFTADS 24A - Maximum altitude with camera lift deployed is FL 1000.</td>
<td>03/08/2022</td>
<td>Active</td>
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</tbody>
</table>
List of EASA Supplemental Type Certificates

| STC Holder                  | Rev. level | Foreign/NAA STC ref. | Movies                                                                 | Description                                                                                           | Certification basis                                                                                      | Associated Technical Documentation                                                                                   | Limitations/Conditions                                                                                     | Date of Issue | Status  | System Record |
|-----------------------------|-----------|----------------------|----------------------------------------------------------------------|--------------------------------------------------------------------------------------------------------|----------------------------------------------------------------------------------------------------------|---------------------------------------------------------------------------------------------------------------|----------------|----------|---------------|
| AEROERVIS, S.R.O.           |           |                      | L 410 UV-P-E, L 410 UV-P-ES, L 410 UV-P-20, L 410 UV-P-E             | The Certification Basis for the original product as amended by the following additional or alternative avionics requirements:  
|                             |           |                      |                                                                      |                                                                                                        |                                                                                                          | None                                                                                                           | 18/03/2020 | Active  | 1072345       |
| BEL TEXTRON, Inc.           | AS 355 F1 | SH1385SSO            | 206L-1, 206L-3, 206L-4, 209A, 206B, 206L                         | The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SH1385SSO. The requirements for environmental protection and the associated certified noise/energy emissions levels of the product remain unchanged and remain applicable to this certificate/approval without any impact. | Master Document List: AE133000X, Rev. P, dated 13.2.2020-Compliance Checklist, AE133001P, Rev. K, dated 20.1.2020-Definitive document, AE133001P, Rev. N, dated 30.1.2020-Review of the above listed document(s) approved/accepted under the EASA system. | - Certification Program doc. n. P92P00100-Csp-05, issue 5, dated 03.03.2020-Compliance Check List doc. n. P92P00100-CLL-03, issue 3 dated 03.03.2020, Master Drawing List doc. n. P92P00100-Mdl-01, issue 1 dated 27.01.2020-Flight Manual Supplement doc. n. P92P00100-FM-02 issue 2, ICA doc. n. P92P001. | 19/03/2020 | Active  | 1072846       |
| SKANDIA, INC                | AS 355 F1 |                      | 2030, B200C, B200GT, B200T, B200, 2020C, B300CCT, B300CT          | The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SAU421CH. The requirements for environmental protection and the associated certified noise/energy emissions levels of the product remain unchanged and remain applicable to this certificate/approval without any impact. | Master Document List: AE133000X, Rev. P, dated 13.2.2020-Compliance Checklist, AE133001P, Rev. K, dated 20.1.2020-Definitive document, AE133001P, Rev. N, dated 30.1.2020-Review of the above listed document(s) approved/accepted under the EASA system. | - Certification Program doc. n. P92P00100-Csp-05, issue 5, dated 03.03.2020-Compliance Check List doc. n. P92P00100-CLL-03, issue 3 dated 03.03.2020, Master Drawing List doc. n. P92P00100-Mdl-01, issue 1 dated 27.01.2020-Flight Manual Supplement doc. n. P92P00100-FM-02 issue 2, ICA doc. n. P92P001. | 19/03/2020 | Active  | 1072846       |

Precautions: This list contains a selection of Supplemental Type Certificates (STCs) issued by the European Union Aviation Safety Agency (EASA). The information provided includes the STC holder, active status, movies related to the STCs, description of the STCs, certification basis, associated technical documentation, limitations/conditions, date of issue, status, and system record. Each STC is associated with specific aircraft models, such as Airbus A319, A320, and A321, and includes details on the installation of various avionics, environmental protection, and noise reduction systems. The STCs cover a wide range of aircraft models, including those manufactured by Boeing, Airbus, and Bell Textron, among others. The certification process ensures that these modifications do not compromise the aircraft's airworthiness and environmental compliance. The date ranges specified (e.g., 27/03/2020) indicate the period during which these STCs were active. The list is designed to provide a comprehensive overview of the active STCs and their associated details, allowing for informed decision-making in the aircraft modification and certification process.
<table>
<thead>
<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NA A STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>10072893</td>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td></td>
<td></td>
<td>737-800</td>
<td>Alternative Maximum Takeoff Weight: Modification of airplane maximum take-off weight to 78,000 kg and maximum landing weight to 66,360 kg.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection are unchanged but this certificate/approval involves a change with an impact on the noisedatabase and/or emissions levels.</td>
<td>Certification Program: MODE00269_CP 19072893</td>
<td>- Applicable to Boeing 737-800 aircrafts, equipped with winglets and CTM-750/782BE engines, APM option code &lt;286951&gt;, having EASA STC 100715798x EASA STC 10071731 embodied. The aircraft must be operated as per basic APM D631A001. Prior to installation of this change/repair it must be determined that...</td>
<td>26/03/2020</td>
<td>Active</td>
<td>10072893</td>
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<tr>
<td>10072895</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>REV. 2</td>
<td>A320-231/-232/-233, A320-211/-212/-214/-215/-219, A320-211</td>
<td>Airbus A320 – Introduction of GLU-925 MRRxRevisions: 1. Transfer of STC.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>ATL AFM Supplement: 13405-FAA-00001 Issue 5A TL 13405-LCA 00001 Issue 14TL 13405-MDL 00001 Issue 5 for later revisions of the above listed document(s).</td>
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<td>10072897</td>
<td>AEROSPACE DESIGN &amp; COMPLIANCE LLC</td>
<td></td>
<td>SEE FAA AML</td>
<td>Installation of the KFC 230 Aerocruise Automatic Flight Control System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC. Requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>(1) Aerospace Design &amp; Compliance, LLC. Master Data List (MDL) 8890019-025, Rev IR, dated 10 July 2019, or later FAA approved revision (2) Aeronautical Design &amp; Compliance, LLC. Instructions for Continued Airworthiness (ICA) 89800019-024, Rev IR, dated 5 July 2019, or later FAA approved revision. (3) Aerial...</td>
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<td>10072904</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST03774AT-D</td>
<td>GULFSTREAM G150</td>
<td>Installation of a Rockwell Collins FMS-6100 Version 4.0 Upgrade inconsistent with Gulfstream Aerospace STC No. ST03774AT-D.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Index List CEC11188001. Revision AF/Flight Manual Supplement CEC11188002.</td>
<td>Rockwell Collins GPS-4000S installation by STC ST0433999-00/01 (EASA10016518) is required. Rockwell Collins DBU-5000 installed by STC ST1037259-DEASA/IMA IL 01 (DBU-5000 is required for serial numbers 201 thru 234. (DBU-5000 is existing in serial numbers 235 and subsequent). Prior to...</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10072906</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST04208AT-D</td>
<td>GULFSTREAM 200/210</td>
<td>JET, G200, G210</td>
<td>Installation of a Rockwell Collins FMS-6100 Version 4 I Upgrade inaccordance with Gulfstream FMS-6100 Version 4.1 Upgrade Installation of a Honeywell ASPIRE 200</td>
<td>Index List CEB11188003, Revision C Flight Manual Supplement CEBS11188005, Revision C Instructions for Continued Airworthiness CEB11189002, Revision C</td>
<td>This STC is only applicable for the following aircraft: (Rockwell Collins avionics equipped aircraft) Prior or concurrent installation of the Rockwell Collins GPS-4000 Global Positioning System by STC ST01433W-D (EASA 10016518)</td>
<td>27/03/2020</td>
<td>Active</td>
<td>10072906</td>
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<td>10072907</td>
<td>HONEYWELL INTERNATIONAL, Inc.</td>
<td>FAA STC SR02667SE</td>
<td>S-76A, S-76B, S-76C, S-76CD</td>
<td>S-76, S-76A</td>
<td>Installation of a Honeywell ASPIRE 200 Intermediate Gain (IG) SATCOMSystem</td>
<td>The Certification Basis for the original product remains applicable to this certificate/-/ or approvals.</td>
<td>- Master Drawing List (MDL) MOD- S76A/400000005-5301, Revision, dated 29 September 2019, or later FAA approved revision - Instructions for Continued Airworthiness (CA) MODS76A/400000005-7501, Revision A, dated 24 July 2019, or later FAA approved revision - Rotorcraft Flight Manual Supplement (RFMS) MOD-</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any oth...</td>
<td>30/03/2020</td>
<td>Active</td>
<td>10072907</td>
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<tr>
<td>10072908</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>FAA STC SR02676SE</td>
<td>MBB-BK117 D-2, MBB-BK117 C-2</td>
<td>MBB-BK117</td>
<td>Installation of Heavy Duty Wearplates</td>
<td>The Certification Basis for the original product remains applicable to this certificate/-/ or approvals.</td>
<td>- Installation Instructions IN-D-145-1069, Revision A, dated 23 August 2019, or later FAA approved revision - Master Document List (MDL) MOD-D-145-1069, Revision B, dated 19 December 2019, or later FAA approved revision - Instructions for Continued Airworthiness (CA) MOD-D145-1069, Revision 0, dated 28 August 201...</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change or/and repair will introduce no adverse effect on the airworthiness of the product.</td>
<td>30/03/2020</td>
<td>Active</td>
<td>10072908</td>
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</table>
The Certification Basis for the original product as amended by additional or later amendments.

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval without any interference.

Installation of ADS-B & LPV Upgrades in accordance with Scandinavian Avionics Modification Ref 2121024.

Approval of ADS-B & LPV Upgrades in accordance with Scandinavian Avionics Modification Ref 2121024.

The Certification Basis for the original product as amended by additional or later amendments.

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval without any interference.

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval without any interference.

The Certification Basis for the original product as amended by additional or later amendments.

The Certification Basis for the original product as amended by additional or later amendments.

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The Certification Basis (CB) for the original product remains applicable to this certificate/ approval without any interference.

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The Certification Basis for the original product as amended by additional or later amendments.
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<tr>
<th>List of EASA Supplemental Type Certificates</th>
<th>easa.europa.eu</th>
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</thead>
</table>

**ATL Modification XL46MDD001: Portable Electronic Flight Bag(EB) armament and power supply activation.**

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged, and remain applicable to this certificate/approval without any impact.

**Altdesignation XL46MDD001, **Portable Electronic Flight Bag(EB) armament and power supply activation.**

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change with impact on the noisedatabase and/or emission levels. The environmental protection requirements are amended as follows: CS-36 Amdt 5.

**Hartzell Propeller Inc. STC for Cirrus SR22T**

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change with impact on the noisedatabase and/or emission levels. The environmental protection requirements are amended as follows: CS-36 Amdt 5.

**Hartzell Propeller Inc.**

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change with impact on the noisedatabase and/or emission levels. The environmental protection requirements are amended as follows: CS-36 Amdt 5.

**Haeco Cabin Solutions, LLC.**

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change with impact on the noisedatabase and/or emission levels.

**Haeco Cabin Solution, LLC.**

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change with impact on the noisedatabase and/or emission levels.

**Mediterranean Aviation Company Ltd.**

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change with impact on the noisedatabase and/or emission levels.

**JAMCO America, Inc.**

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. This certificate/approval involves a change with impact on the noisedatabase and/or emission levels.

**JAMCO America, Inc.**

**JAMCO America, Inc.**

**JAMCO America, Inc.**
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<th>EASA STD No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>10072907</td>
<td>ARMSTRONG AEROSPACE, Inc.d/b/a ASTRONICS ARMSTRONG AEROSPACE</td>
<td></td>
<td>FAA ST04318CH</td>
<td>777-300 777-300ER 777-200 777-200LR</td>
<td>Installation of an Astronics Flight Deck Outfit w/ USB Power System.</td>
<td>The Certification Basis for the original product remains applicable, except where amended by additional or later amendments: CS25 at amdt 02.05.26, 25.303, 25.561(c), 25.571(a)(ii), 25.601, 25.605(a), 25.609(a), 25.611, 25.613(a), 25.619, 25.625(a), 25.777(a).</td>
<td>1. Astronics Armstrong Aerospace Master Data List. Document #: DC10-976-D0-00, Revision C, dated 15 June 2018&lt;sup&gt;1&lt;/sup&gt; (&lt;sup&gt;2&lt;/sup&gt;) or later FAA approved revision. 2. Flight Manual Supplement&lt;sup&gt;2&lt;/sup&gt; (&lt;sup&gt;3&lt;/sup&gt;, 4) Astronics Armstrong Aerospace Document No DC305-976-00-00, Revision (&lt;sup&gt;4&lt;/sup&gt;) or approved 25 June 2018&lt;sup&gt;1&lt;/sup&gt; (&lt;sup&gt;2&lt;/sup&gt;) or later FAA,..</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will not introduce a hazard upon the airworthiness of the product.</td>
<td>21/04/2020</td>
<td>Active</td>
<td>10072907</td>
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<tr>
<td>10073000</td>
<td>SWS CERTIFICATION SERVICES Ltd.</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>787-9</td>
<td>CHINA EASTERN 787-9 VANTAGE XL 734 BUSINESS CLASS SUITE DOOR ACTIVATION This design change certifies the re-activation and use of the suitcases in the closed position during flight conditions only.</td>
<td>The Certification Basis for the original product remains applicable, except where amended by additional or later amendments: CS25 at amdt 02.05.26, 25.303, 25.561(c), 25.571(a)(ii), 25.601, 25.605(a), 25.609(a), 25.611, 25.613(a), 25.619, 25.625(a), 25.777(a).</td>
<td>SWS Certification Services Ltd Master Configuration Scheme No.3209-MCS-01 Issue 3 dated 23 April 2020&lt;sup&gt;1&lt;/sup&gt; or later revisions of the above listed document(s) approved/accepted by EASA system.</td>
<td>Boeing Master Change No. 25205P2259555 will require EASA approval. The STC installed are to be added to an EU register. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce a hazard upon the airworthiness of the product.</td>
<td>24/04/2020</td>
<td>Active</td>
<td>10073000</td>
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<tr>
<td>10073006</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>FAA ST04328AT-D</td>
<td>GV, GV-SP</td>
<td>Installation of Kufra Fin Cap Radome in accordance with Gulfstream Aerospace Corporation Index List CES25190201 Revision (-), dated August 29, 2019 which Section 1 adds the FAA STC reference.</td>
<td>The Certification Basis (CB) for the original product remains applicable, except where amended by additional or later amendments: CS25 at amdt 02.05.26, 25.303, 25.561(c), 25.571(a)(ii), 25.601, 25.605(a), 25.609(a), 25.611, 25.613(a), 25.619, 25.625(a), 25.777(a).</td>
<td>Instructions for Continued Airworthiness, CES25124001 Revision (-), dated July 30, 2019&lt;sup&gt;1&lt;/sup&gt; or later revisions of the above listed document(s) approved/accepted by EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce a hazard upon the airworthiness of the product.</td>
<td>15/04/2020</td>
<td>Active</td>
<td>10073006</td>
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<td>10073008</td>
<td>AEROSPACE DESIGN &amp; COMPLIANCE LLC</td>
<td>SEE FAA APPROVED MODEL LIST</td>
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<td>Installation of Bendix King KA 310 Autopilot Adapter.</td>
<td>The Certification Basis for the original product remains applicable, except where amended by additional or later amendments: CS25 at amdt 02.05.26, 25.303, 25.561(c), 25.571(a)(ii), 25.601, 25.605(a), 25.609(a), 25.611, 25.613(a), 25.619, 25.625(a), 25.777(a).</td>
<td>1. Aerospace Design &amp; Compliance, LLC. Master Data List AML23, MDL-0048, Rev A, dated 18 July 2019, or later FAA approved revision. 2. Aerospace Design &amp; Compliance, LLC. Instructions for Continued Airworthiness (ICA) AML23-ICA-0048, Rev IR, dated 15 July 2019, or later FAA approved revision 3. Aerospace.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed change and/or repair will not introduce a hazard upon the airworthiness of the product.</td>
<td>15/04/2020</td>
<td>Active</td>
<td>10073008</td>
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<td>10073021</td>
<td>JETARE AEROSPACE, LLC</td>
<td>FAA ST03450NY</td>
<td>737-900ER 737-700, 737-800, 737-900 737-400, 737-500, 737-600 737-200, 737-200C, 737-300</td>
<td>Installation of Invicta Mitigation/ Flammability Reduction Technology in the Boeing 737 Center Wing Tank.</td>
<td>The Certification Basis for the original product remains applicable, except where amended by additional or later amendments: CS25 at amdt 02.05.26, 25.303, 25.305, 25.306, 25.307 (a), 25.561(c)(1)), 25.571(b),...</td>
<td>Jatere Aerospace Master Data List. Report No. 1302202012, revision date 03 March 2018 Revision G or later FAA approved revision. Jatere Aerospace Supplement to Instructions for Continued Airworthiness Report No. 1302200504 revision date 07 September 2018 Revision IR or later FAA approved revision a..</td>
<td>The aircraft must be operated in accordance with the “Airworthiness, limitations” section of Jatere Aerospace Supplement to Instructions for Continued Airworthiness Report No. 1302200504 revision 03 May 2018 Revision IR or later FAA approved revision a..</td>
<td>24/04/2020</td>
<td>Active</td>
<td>10073021</td>
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<tr>
<td>10073051</td>
<td>FALCON CREST AVIATION SUPPLY, INC.</td>
<td>FAA STC SR06916R</td>
<td>BD105 S BD105 C</td>
<td>Installation of a Concorde RG390E valve regulated sealed lead-acid battery.</td>
<td>The Certification Basis (CB) for the original product remains applicable, except where amended by additional or later amendments: CS25 at amdt 02.05.26, 25.303, 25.561(c), 25.571(a)(ii), 25.601, 25.605(a), 25.609(a), 25.611, 25.613(a), 25.619, 25.625(a), 25.777(a),.</td>
<td>1. Falcon Crest Ref. drawing number 1 Rev. 1, dated 19 June 1995, or later FAA approved revision. 2. Rotorkraft Flight Manual Supplement (RFMS), dated 19 June 1995, or later RFMS.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce a hazard upon the airworthiness of the product.</td>
<td>15/04/2020</td>
<td>Active</td>
<td>10073051</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10073059</td>
<td>J.G.A</td>
<td>REV. 2</td>
<td>PC-12/47E</td>
<td>PC-12/47</td>
<td>Seat Slides JG Installation of so-called Seat Slides JG into the cabin of the Pilatus PC-12. This optional feature aims at improving passenger comfort in-flight by enabling a longitudinal translation of the seat.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certifiable noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Seat Slides in accordance with J.G. Aviation documents JG001-B Master Drawing List, at Edition 4, dated August 02, 2021 Aircraft Flight Manual Supplement, JG001 Flight Manual Supplement, Rev.0, dated April 01, 2020 Maintenance Manual Supplement, JG001 Maintenance Manual Supplement(s)</td>
<td>11/10/2021</td>
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<td>10073059</td>
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<tr>
<td>10073061</td>
<td>ARMSTRONG AEROSPACE Inc.d/b/a ASTRONAUTICS ARMSTRONG AEROSPACE</td>
<td>FAA STC ST0478CH</td>
<td>A330-300/-300/301/303/304/305/306/307/308/309/310/311/312/313/314/315/316/317/318/319/320/321/322/323</td>
<td>Installation of an Astronics Flight Deck Outlet w/ USB Power System</td>
<td>EASA TCDS level amended by the following requirements: CS 25amd0 (3), 25.303, 25.561 (c), 25.571 (b), 25.601, 25.605 (d), 25.709 (e), 25.813 (f), 25.810 (f)(3), 25.855 (a), 25.771 (a), 25.786 (a), 25.869 (a) (4), 25.951 (a), 25.130 (a)(b)(c)(d), 25.130 (c), 25.130 (a)(b)(c)(d) (g), 25.135...</td>
<td>1. Astronics Armstrong Aerospace Master Data List, Document No. DC104-8847 B-00, Revision E, dated 3 May 2019, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/05/2020</td>
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<td>10073071</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>REV. 1</td>
<td>MYSTERE FALCON 900 (MP900)</td>
<td>Installation of Honeywell H1 Digital Electronic Engine Control (DEEC) Revision 1: Transfer from Dassault Aviation to Honeywell International.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certifiable noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any imp...</td>
<td>Drawing List, Garrett Aviation Services document No. 72-0027-001, Revision 1, dated 28 August 2001 Airbus Flight Manual Supplement, Garrett Aviation Services Document No. 72-0027-001, Revision 1, dated 31 August 2001 or later revisions of the above listed document(s) approved/accepted criteria of EASA.</td>
<td>If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other...</td>
<td>20/04/2020</td>
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<td>10073072</td>
<td>WINGLET TECHNOLOGY, LLC</td>
<td>FAA STO1840WI</td>
<td>680 (SOVEREIGN)</td>
<td>Installation of Transitional Winglets, wing structural reinforcement and related systems changes.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The EASA Certification Basis for the addition of Transitional Winglets for wing structural reinforcement, winglet attachment structure, LED position anti-collision ligh...</td>
<td>Installation in accordance with Winglet Technology Master Drawing List (MDL) ref. Document No. CMO-2000, Rev. A, dated May 19, 2017 Operation in accordance with Winglet Technology Airplane Flight Manual Supplement (AFMS) ref. Document No. 68FT-819T-00, Rev. Original, dated May 5, 2017, or Document No. 68FM-82-...</td>
<td>1) It has been determined that the installation of this STC and does not affect the RVSM qualification of the Cessna Model 680 aircraft. 2) The limitations and conditions of TCDS EASA.MA.033 (FAA 700512W) apply except as noted in this STC. 3) It is not...</td>
<td>20/04/2020</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<th>STC Holder Ref.</th>
<th>Rev. level</th>
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<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10073080</td>
<td>MTN. VIEW AVIATION, LLC</td>
<td>FAA SA01120SE</td>
<td>REV. 1</td>
<td>SEE FAA AML</td>
<td>Installation of Door Steward. This revision caters for the change in ownership, from AP Enterprises, LLC to Mt. View Aviation, LLC.</td>
<td>The Type and OSD Certification Bases (CB) for the original product are applicable to this certificate/approval except where amended by additional or later amendments if indicated on the FAA STC SA01120SE. The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Drawings listed on FAA Approved Model List SA01120SE, dated 29/04/2022, or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU/USA Bilateral Agreement. This STC is valid for aircraft types and models approved by EASA and listed on the FAA STC SA01120SE, issued on 6 November 2019, or later FAA approved revision. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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List of EASA Supplemental Type Certificates

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<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>10073110</td>
<td>SIST FLUITGTECHNIK GmbH</td>
<td>KOGDIAM 100</td>
<td>Cartography Aircraft: Installation of a camera system and control console for cartographical observation.</td>
<td>23/04/2020</td>
<td>Active</td>
<td>10707131</td>
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<td>10073112</td>
<td>BEECHCRAFT BERLIN AVIATION GmbH</td>
<td>GVI (G650)</td>
<td>Installation of a new divan on GVI (G650) in accordance with Master/Documents List E/O N° D11-B18-01, Issue 01</td>
<td>23/04/2020</td>
<td>Active</td>
<td>10707132</td>
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<td>10073116</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA SR00516DE</td>
<td>Installation of mounting hardware and approval for the aircraft to be operated by a single pilot from the left crew seat.</td>
<td>24/04/2020</td>
<td>Active</td>
<td>10707116</td>
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<td>10073128</td>
<td>EIROTECH AVIATION SERVICES LIMITED</td>
<td>REV 1 FAA STC</td>
<td>Automatic Dependent Surveillance Broadcast (ADS-B) Out Installation (Revision 1): Transfer of STC Holder from ALOFT AERORIGHTS to EIROTECH AVIATION SERVICES LIMITED</td>
<td>24/02/2021</td>
<td>Active</td>
<td>10707138</td>
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<td>10073150</td>
<td>AV8 PMA, LLC</td>
<td>HAWKER 900XP, HAWKER 800XP, HAWKER 750, HAWKER 800, BAE 125 SERIES 800B, BAE 125 SERIES 800A</td>
<td>EASA Validation of FAA Parts Manufacturer Approval (PMA) PG88597/Supplement No. 6 dated 10/01/2019 for MLG Pylon Tube Article No. NP-8UM-6, NP-8UM-8, NP-8UM-12, NP-8UM-16, 25-15857 for the requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchangeable remain applicable to this certificate/approval without any im...</td>
<td>28/04/2020</td>
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<td>STC Holder</td>
<td>AV8 PMA, LLC</td>
<td>STC No.</td>
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<td>HS 125 SERIES 700A/700B</td>
<td>HS 125 SERIES 401B/600A/600B</td>
<td>EASA STC No. 8130-3</td>
<td>Produced by the holder of the FAA Form 8130.3</td>
<td>4.3.1.1</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any in...</td>
<td>EASA STC No. 8130-3: “Produced by the holder of the FAA Form 8130.3.”</td>
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<td>AIRCRAFT DESIGN CERTIFICATION GmbH</td>
<td>EA 300/CL EA 300A</td>
<td>EASA STC No. 8130-2</td>
<td>Produced by the holder of the FAA Form 8130.3</td>
<td>4.3.1.1</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any in...</td>
<td>EASA STC No. 8130-2: “Produced by the holder of the FAA Form 8130.3.”</td>
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<td>SAFRAN ENGINEERING SERVICES GmbH</td>
<td>CL-600-2D4 (RJ SERIES 600)</td>
<td>EASA STC No. 8130-1</td>
<td>Produced by the holder of the FAA Form 8130.3</td>
<td>4.3.1.1</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any in...</td>
<td>EASA STC No. 8130-1: “Produced by the holder of the FAA Form 8130.3.”</td>
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<td>TELAR INTERNATIONAL AB</td>
<td>ERJ 190-300</td>
<td>EASA STC No. 8130-0</td>
<td>Produced by the holder of the FAA Form 8130.3</td>
<td>4.3.1.1</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any in...</td>
<td>EASA STC No. 8130-0: “Produced by the holder of the FAA Form 8130.3.”</td>
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<td>AERODATA AG</td>
<td>B300</td>
<td>EASA STC No. 8130-0</td>
<td>Produced by the holder of the FAA Form 8130.3</td>
<td>4.3.1.1</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise level and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any in...</td>
<td>EASA STC No. 8130-0: “Produced by the holder of the FAA Form 8130.3.”</td>
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<td>10073186</td>
<td>WAYPOINT AEROTEC LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>EMB-505</td>
<td>Replacement of Passenger Seat Cushion Covers and Foam</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged, remain applicable to this certificate/ approval without any impact. Installation in accordance with Waypoint Aeroset modification nAFM-678 and Master Data List (MDL) ref WAL-MDL-076 Iss1, dated 25 September 2020. The requirements for continued airworthiness/CAI are unchanged and remain applicable to the EASA system. Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10073189</td>
<td>DIAMOND AIRCRAFT INDUSTRIES GmbH</td>
<td>REV. 1</td>
<td>DA 62</td>
<td>T/O and LDG with RAAC OFF Amended</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged, remain applicable to this certificate/ approval without any impact. Installation in accordance with DHC-8-401 primary navigation display (EHSI) and certify an approved documentation for airplanes without air conditioning system. As a result thereof, no performance reduction during Take-Off compared to an airplane without air conditioning system. Thus same noise level as the airplane...</td>
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<tr>
<td>10073190</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td>B200, B200C</td>
<td>FMS INDICATION ON RH EHSI The purpose of this modification is to design and certify an approved modification for the presentation of the prior installed FMS with RNAV, NAVAID, LPV approval onto the prior installed right hand O3500 primary navigation display (EHSI)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Installation of Honeywell RESCU 406AF/AF&amp; Emergency Locator Transmitter (ELT) on Airbus A319/A320/A321 series aircraft. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Definition in accordance with Engineering Order Doc: 2123900-EO, Issue 1 for the abovementioned document(s) approved/accepted under the EASA system.</td>
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<td>10073230</td>
<td>GOODRICH CORPORATION GE-ICING &amp; SPECIALTY SYSTEMS</td>
<td>FAA STC ST04311CH</td>
<td>DHC-8-402, DHC-8-401</td>
<td>Installation of Goodrich de-icer boots on the engine Inlet</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of thrachtional product are unchanged and remain applicable to this certificate/approval. Goodrich Corporation Drawing 5DS277, Revision B, dated 22 January 2018, or later FAA approved revisions.</td>
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<td>EASA STC no.</td>
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<td>10073230</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>10073235</td>
<td>10073230</td>
<td>SA 315 B</td>
<td>Fabrication of Onboard Systems Model 200-036-00, 200-036-01(+/-) and 200-038-00 and 200-038-01 Load Weigh Kits.</td>
<td>The Certification Basis for the original product remains applicable if certificated equipment is approved, except where amended by additional or later supplements if indicated on FAA STC.</td>
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<td>10073235</td>
<td>TROTTER CONTROLS, INC.</td>
<td>10073236</td>
<td>10073236</td>
<td>AT-802/AT-802A</td>
<td>Fabrication of Fire Retardant Dispersal System, GEN III.</td>
<td>The Certification Basis for the original product remains applicable if certificated approval, except where amended by additional or later supplements if indicated on FAA STC.</td>
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<td>10073236</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>10073294</td>
<td>10073294</td>
<td>P.68 &quot;OBSERVER 2&quot;</td>
<td>STC-AT-148. Rear hatch LASER scanner activation</td>
<td>The Certification Basis for the original product as amended by the following: additional or alternative airworthiness requirements. The following paragraphs shall apply to this change.</td>
</tr>
<tr>
<td>10073294</td>
<td>BIZJET INTERNATIONAL SALES AND SUPP Inc.</td>
<td>10073298</td>
<td>10073298</td>
<td>FAA ST0981AC</td>
<td>737-800 Installation of an Executive Style Interior</td>
<td>The Certification Basis for the original product as amended by the following: additional or alternative airworthiness requirements. The following paragraphs shall apply to this change:</td>
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<tr>
<td>10073298</td>
<td>STC TWENTY ONE LIMITED</td>
<td>10073298</td>
<td>10073298</td>
<td>CAA UK RESPONSIBILITY</td>
<td>727-200F 727-200 ACTIVATION OF RNP-APCH TO LNAV MMIN/WATHS Change is required to provide airworthiness approval of an existing globa...</td>
<td>The Certification Basis for the original product as amended by the following: additional or alternative airworthiness requirements. The following paragraphs shall apply to this change:</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC ref.</td>
<td>Model</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
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<td>10073323</td>
<td>JET AVIATION AG</td>
<td></td>
<td>CL-600-2B19 (RU SERIES 100)</td>
<td>Master Minimum Equipment List (MMEL) Supplement for Challenger 850 JBSL-S-03527MMEL-001 This MMEL Supplement introduces appropriate updates to aircraft dispositional conditions of the EASA MMEL for CL-600-2B19 concerning ATA 52 cabin interior door indication system.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: EASA CS-MMEL MMEL 050 MMEL 105 MMEL 107 MMEL 110 MMEL 115 MMEL 120 MMEL 125 MMEL 130 MMEL 135 MMEL 140 MMEL 145.</td>
<td>Jet Aviation AG MMEL Supplement for Challenger 850 JBSL-S-03527MMEL-001 Issue A or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
</tr>
<tr>
<td>10073324</td>
<td>ESG</td>
<td></td>
<td>EC635 T2, EC635 T3</td>
<td>Installation of a cabin WiFi hotspot with 3G/4G cellular modem and outside antennas in accordance with List of Compliance Documents:00049-DO-M2-LCD-000</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any amendment(s).</td>
<td>- Rotorcraft Flight Manual Supplement 00049-DO-M2-FME-000 Issue 02: Instructions for Continued Airworthiness 00049-DO-M2-ICA-000 Issue 02: List of Compliance Documents 00049-DO-M2-LCD-000 Issue 02: later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<tr>
<td>10073325</td>
<td>INNOTECH AVIATIONA DIVISION OF IMP GROUP LTD</td>
<td></td>
<td>TC6 STC Q-LISA06-043D ISSUE 3</td>
<td>Miscellaneous Interior Modifications, including 1) Minor Galley Modification (addition of removable shelf Slide Lock Modification( + )/ + + ) Two additional L/C/ISA 115 VAC 60 Hz electrical outlets installed -Galley, RH5+ (+ / + + ) Tow Bar Bracket Relocation (+ / + + ) Lockable Compartment Inlet AFT Wardrobe...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any amendment(s).</td>
<td>Master Data List, Kmllex Aviation Inc. document No. MLD26140002, Revision B, dated 10 September 2019 Instructions for Continued Airworthiness, Kmllex Aviation Inc. document No. ICA26140-003, Revision NC, dated 10 September 2019: later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<tr>
<td>10073338</td>
<td>AVIONITEC AG</td>
<td></td>
<td>340A</td>
<td>Avionics Upgrade: Installation of dual Garmin GNS Transceivers 530/690/695; PN/PV capability, L-3 WVX-500 Stormscope system, Avionyne TAS 600 System, Garmin Flight Stream FS 210 and related antennas.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: EASA AMC 20-28 effective 24/06/2012 for Airworthiness Approval and Operational Criteria Related to Area Navigation for Global...</td>
<td>Installation Manual Garmin GNS 530/690, Document Number 190-00357-02, Revision M, Installation Manual Garmin GNS 430/590, Document Number 190-00356-02, Revision M, Installation Manual MID Continental TA 102, Document Number 9611942-02, Revision &amp; Installation Manual Avionyne TAS 600, Document Number 60082-000,Rev...</td>
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<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>ST04324AT-D</td>
<td>REV 1</td>
<td>FAA STC</td>
<td>A321-272N,-272NX</td>
<td>Installation of Ka-Band Jetwave High Speed Data (HSD) System</td>
<td>The Certification Basis (CB) for the original product as amended by this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the product are unchanged and applicable to this configuration/approval without any intervention.</td>
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<tr>
<td>GULFSTREAM CORPORATION</td>
<td>FAA STC</td>
<td>ST04326AT-D</td>
<td>GVI-G600</td>
<td>Installation of Ka-Band Jetwave High Speed Data (HSD) System</td>
<td>The Certification Basis (CB) for the installation of Ka-Band High Speed Data System.</td>
<td>Instructions for Continued Airworthiness Ka-Band High Speed Data – EWIS</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>ATR 72-212 A</td>
<td>REV 1</td>
<td>FAA STC</td>
<td>ATR 42/72 “600 version”</td>
<td>Installation of Ka-Band Jetwave High Speed Data (HSD) System</td>
<td>The Certification Basis (CB) for the original product as amended by this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the product are unchanged and applicable to this configuration/approval without any intervention.</td>
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<td>GULFSTREAM CORPORATION</td>
<td>FAA STC</td>
<td>ST04324AT-D</td>
<td>GVI-G600</td>
<td>Installation of Ka-Band Jetwave High Speed Data (HSD) System</td>
<td>The Certification Basis (CB) for the original product as amended by this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions level of the product are unchanged and applicable to this configuration/approval without any intervention.</td>
<td>Instructions for Continued Airworthiness Ka-Band High Speed Data – EWIS</td>
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<td>EASA STC no.</td>
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<td>10073356</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
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<td>DV 20, DV 20 E</td>
<td>TRACKING DEVICE INSTALLATION</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-VLA paragraphs 13, 32, 101, 301, 303, 305, 307, 321, 337, 341, 471, 561, 572, 601, 603, 695,607, 609, 611, 613, 615, 61...</td>
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<td>10073358</td>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD</td>
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<td>TCCA STC SA18-14 ISSUE 2, BD-700-1A10</td>
<td>KA Band System Installation</td>
<td>The Certification Basis for the original product as applicable lists certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10073359</td>
<td>BOEING IRELAND LIMITED</td>
<td>0</td>
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<td>787-8, 787-9, 787-10, 777-200-200LR/-300/-300ER, 767-200-300/-300ER, 737-700/-800/-900, -90ER, 737-800/-900/-900/-909-900ER</td>
<td>Zero Occupancy in Passenger Cabin/Installation 1 adds Boeing 737-8 and 737-9 aeroplane models into the applicability.</td>
<td>The Certification Basis for the original product as applicable lists certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
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<td>10073363</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>0</td>
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<td>CL-600-2B19 (RJ SERIES 100)</td>
<td>REAR PRESSURE BULKHEAD FEED THRUMechanical installation of several electrical connectors and reinforcement at rear pressure bulkhead on Bombardier Challenger CL-650</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10073367</td>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>0</td>
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<td>FFA ST03506NY</td>
<td>MYSTERE FALCON 900 (MF900)</td>
<td>Honeywell FMS CD-830 Displays UpgradeRemoval of existing Honeywell CD-810 or CD-820 Control Display Units (CDUs) and replaced with Honeywell CD-830 CDUs</td>
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<tr>
<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>A320-212N/-272NX A321-271N/-271NX/-271NX X A321-233N/-253N/-252N X A321-213/-232/-232/-281N/-252N A321-111/-112/-113/-211/-212 A320-253N/-271N/-271N/-271N A320-231/-232/-232/-232/-231N X A320-211/-212/-214/-215/-216 A319/-171N A219/-131/-132/-133/-151N A19/-111/-112/-113/-114/-115</td>
<td>Installation of Stretch NRG on A320 Family</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Surveillance - Broadcast (ADS-B) Out optional Automatic</td>
<td>The requirements for Environmental Protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval. The requirements for Environmental Protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>20/05/2020</td>
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<tr>
<td>AVIADECO GmbH</td>
<td>A320-214</td>
<td>Cabin Re-Configuration: This modification consists of a cabin reconfiguration to the maximum allowable passenger seat configuration of Y180 for A320 aircrafts.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td></td>
<td></td>
<td>10/06/2020</td>
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<tr>
<td>AEROMECH Inc.</td>
<td>FAA STC SA0254SE</td>
<td>PA-46-600TP PA-46-500TP</td>
<td>Installation of Stretcher NG on A320</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/07/2020</td>
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<td>KAIERAIR, INC.</td>
<td>FAA STC ST0342CH</td>
<td>GV GV</td>
<td>Installation of an Universal Avionics UNS-1E Pw flight management systems with Link XL-801 datalink system and CVR-120A Cockpit voice recorder; meet airworthiness requirements for FANS 1A+ and optional Automatic Dependent Surveillance - Broadcast (ADS-B) Out system.</td>
<td></td>
<td></td>
<td>03/06/2020</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
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<td>Foreign/NAA STC ref.</td>
<td>Model</td>
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<td>10073398</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>3</td>
<td>A340-311/312/211/212-213</td>
<td>HAWKER 900XP, WATTUM, JOEL</td>
<td>Installation of Aerocet 2200 floats</td>
<td>The Certification Basis for the original product as amended/modified includes additional or alternative airworthiness requirements - Special Condition (s): F-108 Security Protection of Aircraft Systems and Networks. The requirements for environmental protection and the associated notified noise and/or emissions levels of the product remain applicable to this certificate/approval without any impact.</td>
</tr>
<tr>
<td>10073399</td>
<td>AVS PMA, LLC</td>
<td>3</td>
<td>HAWKER 900XP, HAWKER 750, 850</td>
<td>EASA Validation of FAA Parts Manufacturer Approval (PMA) P03038855/010/10 issued 03/17/2020 for Article NP-UM153A-1 “Bolt Assembly” as approved replacement for Article Number 25UM153A and NP-UM153A-3 “Bolt” as approved replacement for Article Number 25UM153-1n Tastron Hawker Series listed above.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated notified noise and/or emissions levels of the product remain applicable to this certificate/approval without any impact.</td>
<td></td>
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<tr>
<td>10073400</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>3</td>
<td>A330-343</td>
<td>Removal of Ka-Band Satcom System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated notified noise and/or emissions levels of the product remain applicable to this certificate/approval without any impact.</td>
<td></td>
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<tr>
<td>10073403</td>
<td>WATTUM, JOEL WILLIAM AND ORD HARE- WATTUM BARBARA JEANNE</td>
<td>3</td>
<td>FAA STC SA024000AK</td>
<td>SEE FAA AML</td>
<td>Installation of Aerocost 2200 floats</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA024000AK. The requirements for environmental protection and the associated notified noise and/or emissions levels of the product remain applicable to this certificate/approval without any impact.</td>
</tr>
<tr>
<td>10073407</td>
<td>SWIFT AIRCRAFT LTD</td>
<td>3</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY G 115E</td>
<td>Consipicuity Wrapping The UK MoD requires a Complimentary Assessment of the Grob Tutor G115E aircraft that they currently use for AFS (Air Experience Flights) and US (University Air Squads) under a previous EASA STC 10073407. Swift aircraft obtained permission for wrapping with a certain manufacturer.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated notified noise and/or emissions levels of the product remain applicable to this certificate/approval without any impact.</td>
</tr>
</tbody>
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Additional notes:
- The change is defined with Revision D of Master Document List document ADDL- A3401-1-4430. The Aircraft shall be operated in accordance with Revision A of Aircraft Flight Manual Supplement document AFMS-A3401-1-4430. Or later revisions of the above listed document(s) approved/accepted under the EASA system.
- AVS PMA, LLC. PMA No.: P03038855/010/10 issued 03/17/2020 for Article Nr. NP-UM153A Bolt (Jacket to pivot), Revision A, dated 4-29-2020. PMA, LLC. Document No. AVS-ALSD0001, Airworthiness Limitation/Supplement Rev. A, dated 2-6-2020 or later revisions of the above listed document(s).
- With the application of this STC the technical documentation A0-1856-FCD, A0-1856-AMF and A0-1856-ICA become obsolete. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherwise preinstalled change/repair and repair will not introduce roadnoise effect upon the airworthiness of the product. This STC is valid for aircraft types and models...
<table>
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<tr>
<th>EASA STC No.</th>
<th>STC holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC no.</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
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<tr>
<td>10073445</td>
<td>MAGELLAN AEROSPACE</td>
<td>TCA SH17-39 ISSUE 1</td>
<td>566</td>
<td>Installation of Wire Strike Protection System.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Modification Data Summary EM10725 Issue C</td>
<td>Flight Supplement Manual Issue PM-965-008 Issue Amendments for Continuing Airworthiness PM-965-006 Issue 4C; later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of E/E Car.</td>
<td>09/06/2020</td>
<td>Active</td>
<td>10073445</td>
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<tr>
<td>10073460</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC ST04465AT</td>
<td>GULFSTREAM G280</td>
<td>Installation of the Concorde Lead Acid Battery RG-380 to replace the OEM Main Lead Acid Battery</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for the environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Master Drawing List: 5-0691, Revision C, dated 16 August 2019, or later FAA approved revision.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any otherwise installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/04/2020</td>
<td>Active</td>
<td>10073445</td>
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<tr>
<td>10073469</td>
<td>T.P.S. S.P.A.</td>
<td>REV. 1</td>
<td>A109E</td>
<td>Right pile with RIELS VP-14 p/n TPS106-03-0049a kit (STC) consists in a pole, installed at the front right side of the helicopter, with a built-in processor and a digital camera for airborne data recording.</td>
<td>The Type and OSG Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Airworthiness Limitation Section: TPS109-0300- ALS-0041Flight Manual Supplement: TPS109-0300-RFMA5-000r later revisions of the above listed document(s) approved/accepted on behalf of EASA.</td>
<td>This approval is limited to the helicopter A109E s/n 11825-TEBA due to the certification process conducted in accordance with the NOHazard Criteria for the impossibility to fully comply with FAR 27.131(a) of the CB.</td>
<td>19/05/2022</td>
<td>Active</td>
<td>10073469</td>
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<tr>
<td>10073482</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>REV. 1</td>
<td>412</td>
<td>EFIS INSTALLATION IN B412FIVE-TUBE BENDIX/MODEL 58F5 ELECTRONIC FLIGHT INSTRUMENTATION SYSTEM</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Master Document List: 41340-RPT02-EC1/ISS 2Flight Manual Supplement 41340-HDK01 EC1/ISS 3Maintenance Manual Supplement 41340-HDK02 EC1/ISS. 2016 later revisions of the above listed document(s) approved/accepted on behalf of EASA.</td>
<td>This Approval is limited to the Helicopter Bell 412 S/N 36022 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2020</td>
<td>Active</td>
<td>10073482</td>
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<tr>
<td>10073485</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>551</td>
<td>AMBULANCE MODULE INSTALLATION IN CESSNA 515</td>
<td>This modification consists of the installation of one ambulance module of PN 2000-018 in passenger compartment. The passenger cabin has been configured, passenger seat number 7 and driver (passenger seats nos. 4, 5 &amp; 6) have been removed in order to allow installation.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>55102-RPT02 Master document list Ed855102-HDK1 Flight manual Supplement Ed85004-000r later revisions of the above listed document(s) approved/accepted on behalf of EASA.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2020</td>
<td>Active</td>
<td>10073485</td>
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<tr>
<td>10073494</td>
<td>DIAMOND AIRCRAFT INDUSTRIES GmbH</td>
<td>REV. 2</td>
<td>DA 62</td>
<td>The aim of this project is to provide an alternative sensor configuration for the DA 62 with more comfort in the 1st passenger row. The 3-seat 1st passenger row is replaced by a 2-seat RH interface seat as provision for 2 seats with an increased distance to the pilot.</td>
<td>The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.</td>
<td>STC 62-0024FM Supplement STC 62-002 to Doc. No. 7.01 25-E and Doc. No. 11.01-06-E later revisions of the above listed document(s) approved/accepted on behalf of EASA.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/07/2020</td>
<td>Active</td>
<td>10073494</td>
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</table>
10073498 ARWOLF FILTER CORP
FAC FAA STC SA0024NY
SEE FAA AML SA0024NY
Installation of Airwolf remote mounted engine oil filter for CFC-1007 or multi-engine fixed wing light aircraft incorporating Lycoming 450 horsepower engines.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.

1. Aircraft Serial Numbers 10073498


10073496 WIPARE Inc.
FAC FAA STC SA0438CH
208B 208
Installation of extended baggage assembly

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval without any change.


10073508 CERTIFYNATION, LLC
FAC FAA STC ST0441CH
A380-861 A380-842 A380-841
Installation of ELT System Upgrade on Honeywell RESCU 406/PN2 Emergency Locator Transmitter (ELT) System Upgrade.

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any change.

1. CertiNation Master Data List doc. no. CN19-044IA0-01 rev 0, dated 04/2020. Certification Instructions for Continued Airworthiness (ICA) doc. no. CN19-044IC0-01 rev 0, or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation


10073509 SAAB AB
REV. 1
35A

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI F-01, The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.

1. Saab AB project TDM02-02 including documents as listed in TDM02-02-LEP Rev:C AFSM-supplement TDM02-02-4-AFSM, Rev:B higher revisions of the above listed document(s) approved/accepted by EASA in accordance with the Technical Implementation.

1. Saab AB project TDM02-02, dated 17 June 2019, or later FAA approved revision.

10073516 S4A SOLUTIONS FOR AVIATION, S.L.
390
WiFi ACTIVATION IN B300/700 design change activates the WiFi system on board Beechcraft 390 aircraft.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS23 paragraphs 23.123, 23.123, 23.21, 23.21, 23.25, 23.25, 23.121, 23.121, 23.123, 23.123, 23.121, 23.121, 23.123, 23.123, 23.121.


10073520 AIR ALLIANCE GmbH
31A 31
Installation of GPS/SBAS Antenna

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: FAR 25.123 (a) amendment 25-123, FAR 25.1316 (a) amendment 25-134, FAR 25.152 (a) amendment 25-54, the requirements for...
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<th>Movie</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>PMV ENGINEERING</td>
<td>REV. 1</td>
<td>737-8, 737-9, 737-800-700-800-900-908ER</td>
<td>Titled crate modification on 8737. The modification consists of the installation of 2 arm mounts on the existing sliding windows (both sides, Captain and First Officer) and the installation of titled crates on those arm mounts.Rev 1 introduces configuration for the installation of the 11 in-estallable...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following paragraphs at a later amendment: CS25 amdt 12:301 (a)(b), 303, 305(a)(b)(c)(x)(v)(y), 307(a)(a), 96(a)(a)(b)(c)(d)...</td>
<td>8 inch tablet crate configuration.: MASTER DOCUMENT LIST, ref: AIV-00200-MLD-04 dated 28/05/2020. INSTRUCTIONS FOR CONTINUED AIRWORTHINESS, ref: AIV-00200-ICA-501. R03 dated 28/05/2020. AIRCRAFT FLIGHT MANUAL SUPPLEMENT. Inset: AIV-00200-AMFS-501-R02 dated 28/02/2020. MMEL Supplement. AIV-00200-MMELS-...</td>
<td>Limitations are defined in the ICA document. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/06/2023</td>
<td>Active</td>
<td>10073520</td>
<td></td>
</tr>
<tr>
<td>REB TECHNOLOGIES Inc./ dba REBTECH</td>
<td>REV. 1</td>
<td>FAA SR09561RC</td>
<td>Installation of NVG Infrared Filter System. This project installs an NVG Infrared Filter System for use with Night Vision Goggles (NVGs) in MD Helicopters Inc. (MDHI), Models 399E, 369FF, 500N to reduce or eliminate infrared emissions detrimental to NVG users.</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: FAA – 14 CFR Part 27.27-1(a)(a). amdt 27.2-27.301(a), 27.301(c). amdt 27.2-27.305, amdt 27.2-27.306(a). amdt 27.2-27.307(a) am...</td>
<td>EASA approved documents.: Rotorcraft Flight Manual Supplement, Document Number: 9820500-ERFMS-Rev. JR dated 21 April 2020. EASA accepted documents.: Master Drawing List. Document Number: 9820500-MLD. Rev. AR dated 27 January 2020. Configuration Control List. Document Number: 9820500-ECCL. Rev...</td>
<td>Rotorcraft certified by this design change employs Night Vision Imaging System lighting, commonly named NVIS lighting.</td>
<td>19/06/2020</td>
<td>Active</td>
<td>10073507</td>
<td></td>
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<tr>
<td>HELICOPTER TECHNOLOGY COMPANY</td>
<td>FAA STC SR02402LA, 07.10.2020</td>
<td>204B, 205A-1</td>
<td>Installation, operation, and maintenance of Helicopter Technology/Company main rotor blade assembly.</td>
<td>The Certification Basis for the original product applies to the following documents, if included on FAA STC SR02402LA.: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>- Master Document List. MDL-204-2100-01, Rev. C, dated 24 January 2018. Instructions for Continued Airworthiness of Maintenance Manuals/HFCTM-006. Rev. D, dated 03 January 2020. Within Airworthiness Limitation Section at Rev. D dated 13 April 2020 for later revisions of the above listed document(s).</td>
<td>Installation of HTC Main Rotor Blade with Part Number 204P0100-101 is not approved under this STC certificate. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/11/2020</td>
<td>Active</td>
<td>10073501</td>
<td></td>
</tr>
<tr>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td>SCT SERIES 3</td>
<td>Installation, operation and maintenance of Electronics International/MP-507 Class Panel Monitor System in SCT-7. &quot;Skyvan&quot; in accordance with Master Document List AIR-2017-024-MLD-01-00, dated 5 June 2020.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25 amdt 4. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>Master Document List: AIR-2017-024-MLD-01-00, dated 5 June 2020. Instructions for Continued Airworthiness Maintenance Manuals/HFCTM-006. Rev. D, dated 03 June 2020. Within Airworthiness Limitation Section at Rev. D dated 13 April 2020 for later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/06/2020</td>
<td>Active</td>
<td>10073504</td>
<td></td>
</tr>
<tr>
<td>QUIT TECHNOLOGY AEROSPACE, Inc.</td>
<td>FAA STC ST04265AT</td>
<td>BAE125 SERIES 1000B, BAE125 SERIES 1000A</td>
<td>Replacement of the inner aluminum barrel of the engine inlet cowl with a Quiet Technology Aerospace inlet barrel of carbon graphite sandwich construction.</td>
<td>The Certification Basis for the original product remains applicable to the following documented approval, except where amended by additional or later amendments if indicated on FAA STC ST04265AT.: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>1. Quiet Technology Aerospace document: &quot;Hawker 1000 Inlet Cowl Inner Barrel Replacement Master Drawing List.&quot; Report No.QTA 6101, Rev. A dated 16 October 2010, or later FAA approved revisions. 2. &quot;Hawker 1000 Inlet Cowl Inner Barrel Replacement Assembly Procedure.&quot; Report No. QTA 6107, Rev. A, dated...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/06/2020</td>
<td>Active</td>
<td>10073503</td>
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<td>STC No.</td>
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<td>Foreign NAA STC no.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10073637</td>
<td>SABENA TECHNICS BGC</td>
<td>A330-300</td>
<td>A330-300 - Cabin Reconfiguration &amp; IFE Modification from 3 classes to 375 PAX Y/C</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval.</td>
<td>- Instructions for continued airworthiness 40533-01-A-ICAO-R00 - Airplane Flight Manual Supplement 40533-01-A-AMS-R00 - Master Data List 40533-01-A-MLD-R00; later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>As a concurrent requirement of this change, accomplishment of this STC requires the simultaneous accomplishment of the following modification:- Installation of a photo-luminescent EFM as per Luftansa Technik EASA STC No. 10014507 Rev.0 Prior to installation of this change/repair it must be determined...</td>
<td>18/06/2020</td>
<td>Active</td>
<td>10073577</td>
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<tr>
<td>10073652</td>
<td>BREEZE EASTERN, LLC.</td>
<td>REV. 1 FAA</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of alternate host cable/Rev. 1: Changed STC Holder.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval.</td>
<td>- FAA STC List 40533-1B-AFM on 737-800/737-900 - 737-800/737-900 - 737-800/737-900: Revision 18/06/2020, or later FAA approved revision 2. Instructions for Continued Airworthiness Refer to FAA AIM for relevant document as applicable to TypeCertificate Holder and helicopter Type and Model Original STC Number: FAA...</td>
<td>18/06/2023</td>
<td>Active</td>
<td>10073582</td>
<td></td>
</tr>
<tr>
<td>10073683</td>
<td>THALES AVIONICS Inc.</td>
<td>REV. 1 FAA</td>
<td>ST2610LA</td>
<td>INSTALLATION OF SATELLITE TELEVISION SYSTEM THALES STV+ ON AIRBUS A320/PLANEThe THALES STV+ system provides cabin live television and stored content to each seatback display. In addition, the system also integrates independent on-ground data link capability for airline operations and maintenance.Rev...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Finding(s)/EASA CRI C-011 Large/High Installation/EASA CRI C-02 Vibration &amp; Bumping. The following paragraph(s) at a later amendment: JAR 25.562 Ch...</td>
<td>- See Inspections in Report 320M8001-1-NCA Rev B Supplemental Instructions for Continued Airworthiness, and EWS Instructions for Continued Airworthiness 320M8001-1-EWS. Revision 1, dated April 23, 2017 - The following Airbus Modifications must be reinstalled prior to conformance with the EASA STC: Mod...</td>
<td>15/07/2020</td>
<td>Active</td>
<td>10073563</td>
<td></td>
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<tr>
<td>10073686</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>CAA UK RESPONSIBILITIES</td>
<td>757-200</td>
<td>Zero Passenger Layout. This Modification is raised to reconfigure the subject aircraft from 452 F/C &amp; 9 Y/C Layout to a Zero Passenger Layout.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval.</td>
<td>- A LIMITATIONS (operational): No passengers on board - No supermariner crew on board - No cargo in passenger cabin B LIMITATION (for Operational Suitability Data Cabin Crew Data (OSDDC)); - OSD CCD is not applicable to this aircraft type. Prior to installation of this change/repair it must be determined...</td>
<td>22/06/2020</td>
<td>Active</td>
<td>10073586</td>
<td></td>
</tr>
<tr>
<td>10073671</td>
<td>SBARS, LLC</td>
<td>FAA STC</td>
<td>DA-40</td>
<td>Installation of a Supercharged Lycoming IO-360-M1A engine</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any impact.</td>
<td>- Master Drawing List No. MDL-D040-1 Rev. R, dated 28 August 2018, or later FAA approved revision 2. Instructions for Continued Airworthiness Document No. 20040-30, Rev. R, dated 15 June 2017, or later FAA revisions.3. Airplane Flight Manual Supplement No. AFMS(20040); Rev. R, dated 20 November 2018.</td>
<td>The supercharged IO-360-M1A engine installation is limited to 30 inhPAP and 2700 RPM. This Supplemental Type Certificate cannot be used unless EASA STC 910073671 has been applied to the IO-360-M1A engine. Prior to installation of this change/repair it must be determined...</td>
<td>26/06/2020</td>
<td>Active</td>
<td>10073711</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10073576</td>
<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>REV. 1</td>
<td>A321-371N-271N-272N</td>
<td>Installation of 90-Day Underwater Location BeaconRev 1: Transfer of STC from GVH Aerospace Ltd - United Kingdom, to GVH Aerospace (Ireland) Limited</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: CS25 Amendment 24:25.1, 25, 29, 25, 301, 25, 305, 25, 307, 25, 321, 25, 337, 25, 341, 25, 345, 25, 349, 25, 350, 25, 361, 25, 367, 25, 369, 25, 383, 25, 1301, 25, 25...</td>
<td>EASA Approved GVH Aerospace Instructions for Continued Airworthiness No.GVH-10387-02-ICA-01</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/01/2021</td>
<td>Active</td>
<td>10073576</td>
</tr>
<tr>
<td>10073577</td>
<td>SBARS, LLC</td>
<td></td>
<td>IO-380-M1A</td>
<td>Installation of SBARS Supercharger system on Boeing 737-800 aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>1. Master Drawing List No. MDL-DA40-1-E, Rev. A, dated 18 August 2018, or later FAA approved revisions. 2. Instructions for Continued Airworthiness, Document No. 20040-31, Rev. IR, dated 13 June 2017, or later FAA approved revisions.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/06/2020</td>
<td>Active</td>
<td>10073577</td>
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<tr>
<td>10073588</td>
<td>INGENIO AEROSPACE INC.</td>
<td></td>
<td>A319-111/-112/-113/-114/</td>
<td>Installation of Pilot's and Co-Pilot's Ingenio Aerospace Tripod Tablet Holder Mounts</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>Modification must be performed in accordance with the following: 1. Krnei Aviation Inc. Master Document List – Installation of INGENIO Cockpit Tripod Mount System on Bombardier Global Express Series Aircraft, Number MDL-DA40-1-E, Revision G, dated 18 June 2020. 2. Ingenio Aerospace Installation Instructions...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/06/2020</td>
<td>Active</td>
<td>10073588</td>
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<tr>
<td>10073594</td>
<td>GAMA AVIATION (ENGINEERING) LIMITED</td>
<td></td>
<td>GAMA-840D-1-1</td>
<td>Installation of provisions for and subsequent installation of Garmin G1000 NXi Integrated Avionics System and GFC 700 Automatic Flight Control System.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25 Amendment 3:25.1, 25, 301, 303, 305, 307, 321, 337, 341, 361, 601, 603, 605, 607, 609, 611, 613, 671, 771; 773; 777;...</td>
<td>Gama Aviation Master Data List, document GAEL-164MDL-01, Issue 2, dated 29 June 2020. Part A: Gama Aviation Drawing GAEL-164-1-01 – General Arrangement. Part B: Garmin Drawing 005-00424-43 – General Arrangement. To be operated according to Gama Aviation Aircraft Flight Manual/Supplement, document GAEL...</td>
<td>This STC is only applicable to aircraft ex-factory (or equivalent equipped) with Collins Pro Line Fusion avionics. The following STCs must be installed concurrently, respectively.</td>
<td>29/06/2020</td>
<td>Active</td>
<td>10073594</td>
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<tr>
<td>10073595</td>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td></td>
<td>ST04422AT</td>
<td>Installation of Gablex Engineering Audio Control Panels</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25 Amendment 11: 25, 1501CS25 and 16, 25, 301, 25, 305, 25, 307, 25, 321, 25, 337, 25, 341, 25, 345, 25, 349, 25, 350, 25, 361, 25, 367, 25, 369, 25, 383, 25, 1301, 25, 251929,...</td>
<td>1. Master Data List (MDL) 217232-MLD-15, Rev. IR, dated 9 January 2019, or later FAA approved revision. 2. Instructions for Continued Airworthiness (ICA) document 217232-ICA Rev. IR, dated 24 October 2017, or later FAA accepted revision.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/09/2020</td>
<td>Active</td>
<td>10073595</td>
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<td>STC Holder</td>
<td>Status</td>
<td>Certification Basis</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>The Certification Basis for the original product as amended by FAA STC ST04462CH.</td>
<td>A310-304</td>
<td>ADS-B Out Activation</td>
<td>- Compliance Checklist CCL-34-19/07M Rev 2 23 June 2020 - Supplemental Type Definition Document ST0204-1907M Issue 1 Rev Dated 22 June 2020 - Aircraft Flight Manual A-00-19/79-98-1FM Issue 02 Rev 00 dated 85 Ox 00-19 - Instructions for Continued Airworthiness A-00-19/79-98-1ICA Issue 01 Rev01 dated 06 J.</td>
<td>This approval is only valid for aircraft which have undergone an SM/GKU according to the procedure BAC-QGES01/P0. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will intro...</td>
<td>25/06/2020</td>
<td>Active</td>
<td>10073596</td>
<td></td>
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<tr>
<td>AIRWOLF FILTER CORP</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate approval without any impact.</td>
<td>A-1</td>
<td>Installation of Airwolf Filter Model AFC-500 Full Flow Engine Oil Filter/P/N ST04462CH.</td>
<td>- 1. Aircraft Installation Drawing List AFC-A001 dated 30 March 1993. 2. Installation Instructions No. AFC-A005-A dated 30 March 1993. 3. Airwolf Maintenance Instructions AFC-A005-M dated 30 March 1993.</td>
<td>Installation is limited to AVIAT A-1 aircraft with Lycoming engine model O-360-CI and O-360-AP Fire wall must be minimum 0.025 inch ASTM A527 galvanized steel one equivalent. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any...</td>
<td>26/06/2020</td>
<td>Active</td>
<td>10073610</td>
<td></td>
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<tr>
<td>SABENA TECHNICS BOC</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate approval without any impact.</td>
<td>ATR 72-212</td>
<td>Installation of Row 44 Broadband Antenna System Provisions</td>
<td>- Master Data List, Ref. 40681-01-A-MDL-R05 dated 20 Dec 2021 - Aircraft Flight Manual Supplement, Ref. 40681-01-A-AMSL-R07 dated 20 Dec 2021 - Instructions for Continued Airworthiness, Ref. 40681-01-A-ICA-R06 dated 20 Dec 2021or later revisions of the above listed document(s) approved or accepted/published...</td>
<td>Limitations as per §3 of Instructions for Continued Airworthiness document and §3 of Aircraft Flight Manual Supplement document referenced above. After 1 year or 2 000 FH threshold after STC amendment, whichever comes first, but no later than 31 July 2022 (see DEV- D25.855-01 Issue 3.1). No cargo is all...</td>
<td>10/01/2022</td>
<td>Active</td>
<td>10073617</td>
<td></td>
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<tr>
<td>ROW 44, Inc.</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate approval, except where amended by FAA STC ST04462CH. The requirements for environmental protection and the associated certified noise and/or emissions levels of the on...</td>
<td>A321-231/A</td>
<td>Installation of Row 44 Broadband Antenna System Provisions</td>
<td>Row 44 Master Data Index Listing, Document No. RWO100-0001-00, Revision RV, dated 4 May 4, 2020 or later FAA approved revision FAA Approved Airplane Flight Manual Supplement, Row 44 Document No. as listed, is required on board the modified aircraft. A319 models: RWO220-0007-00, Revision - approved.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/07/2020</td>
<td>Active</td>
<td>10073636</td>
<td></td>
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<tr>
<td>AIRWOLF FILTER CORP</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate approval, except where amended by FAA STC ST04462CH. The requirements for environmental protection and the associated certified noise and/or emissions levels of the origin...</td>
<td>0-470-J</td>
<td>Installation of oil fitting (AN-832-5D) in rocker box cover</td>
<td>Walker Engineering Company Drawing List No. DL W2031, Revision A, dated May 17, 1976, or later FAA approved revisions.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC has to be used in conjunction with EASA...</td>
<td>06/07/2020</td>
<td>Active</td>
<td>10073652</td>
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<td>Foreign/NAA ID (if any)</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AEROSPACE SYSTEMS INTERNATIONAL LLC</td>
<td></td>
<td></td>
<td>B300, B300C, B200C, B200T B200, B200C, B200CGT</td>
<td>Installation of a remote heading select panel into the pedestal to be used in conjunction with the existing Collins Flight Guidance System (FGS-3000) in a Textron Aviation Inc. B200, B200C, B200CGT (C-12R), B200C (UC-12F), B200CGT (UC-12M), B200C (-C12), B200C (UC-12F), B200CGT (UC-12M), B200C (-C12R) B300C, B300C (...).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC AIRWOLF FILTER AEROTEC &amp; Installation of AD-AFIS-131.</td>
<td>Master Data List: Document No. 632-00-0013, Revision (R), dated 13 November 2019, or later FAA approved revision. FAA approved revision SA12K7DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval without any amendments...</td>
<td>Prior to installation of this change/repair, it must be determined whether this change/repair and any previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This installation requires the installation of Pro Line...</td>
<td>30/06/2020</td>
<td>Active</td>
<td>1073692</td>
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<td>AEROTECA &amp; CONCEPT</td>
<td></td>
<td></td>
<td>ATR 72-212 A, ATR 72-212 ATR 72-211, 212 A, 212 212-201 ATR 72-212 ATR 72-201 ATR 72-102 ATR 72-101 ATR 42-500 ATR 42-500-500 ATR 42-400 ATR 42-320 ATR 42-300 ATR 42-200-300 ATR 42-200</td>
<td>ATR 42/72 Stretcher Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any amendments...</td>
<td>Master Data List AE19140MDL Rev. 0 dated 09/06/2020. Installation is limited to configurations as per drawing STR-16077-001-000 without Medical or Oxygen Equipment. Prior to installation of this change/repair it must be determined whether this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/07/2020</td>
<td>Active</td>
<td>1073699</td>
<td></td>
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<tr>
<td>AERODATA AG</td>
<td>REV. 1</td>
<td></td>
<td>680 (SOVEREIGN)</td>
<td>Installation of AD-AFIS-1310.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any amendments...</td>
<td>Master Document List: Document MDL-AIC-632, Revision 1, dated 29 June 2020. To be operated according to: Airplane Flight Manual Supplement, document AMM-117, Revision 0, dated 18 May 2020.</td>
<td>Limited to S/N 680-5061 and on. Prior to installation of this change/repair, it must be determined whether this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/07/2020</td>
<td>Active</td>
<td>1073699</td>
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<tr>
<td>AIRWOLF FILTER CORP</td>
<td>FAA SE92WE</td>
<td></td>
<td>L110-470-A, 10-470-U, 10-470-V, 10-470-B, 10-470-C, 10-470-D, 10-470-E, 10-470-F</td>
<td>Installation of oil fitting (AN-825-D) in rocker box cover.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any amendments...</td>
<td>Walker Engineering Company Drawing List No. DLW-2031, Revision &quot;A&quot;, dated 06 May 1978. (+) &gt; or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU EASA Bilateral Agreement.</td>
<td>This STC has been used in conjunction with EASA STC 100/16590. Prior to installation of this change/repair it must be determined whether this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/07/2020</td>
<td>Active</td>
<td>1073767</td>
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<tr>
<td>AIRWOLF FILTER CORP</td>
<td>FAA SE93WE</td>
<td></td>
<td>T150-470-D, T150-470-C, T150-470-B</td>
<td>Installation of oil fitting (AN-825-D) in rocker box cover.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any amendments...</td>
<td>Walker Engineering Company Drawing List No. DLW-2031, Revision &quot;A&quot;, dated 06 May 1978. (+) &gt; or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU EASA Bilateral Agreement.</td>
<td>This STC has been used in conjunction with EASA STC 100/16590. Prior to installation of this change/repair it must be determined whether this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/07/2020</td>
<td>Active</td>
<td>1073767</td>
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<td>Active</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>AIRWOLF FILTER CORP, 10073694</td>
<td>AIRWOLF FILTER CORP, 10073694</td>
<td>AIRWOLF FILTER CORP, 10073694</td>
<td>Installation of oil fitting (AN-823-D) in rocker box.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product remain unchanged and are applicable to this certificate/approval without any...</td>
<td>Walker Engineering Company Drawing List No. DLW-2031, Revision “A”, dated 08 May 1978. The following paragraph(s) at a later amendment:</td>
<td>This STC has to be used in conjunction with EASA STC 10016590. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/07/2020</td>
<td>Active</td>
<td>10073694</td>
<td></td>
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<tr>
<td>AIRWOLF FILTER CORP, 10073691</td>
<td>AIRWOLF FILTER CORP, 10073691</td>
<td>AIRWOLF FILTER CORP, 10073691</td>
<td>Installation of oil fitting (AN-823-D) in rocker box.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product remain unchanged and are applicable to this certificate/approval without any...</td>
<td>Walker Engineering Company Drawing List No. DLW-2031, Revision “A”, dated 08 May 1978. The following paragraph(s) at a later amendment:</td>
<td>This STC has to be used in conjunction with EASA STC 10016590. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/07/2020</td>
<td>Active</td>
<td>10073691</td>
<td></td>
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<tr>
<td>DASSAULT FALCON SERVICE, 10073692</td>
<td>DASSAULT FALCON SERVICE, 10073692</td>
<td>DASSAULT FALCON SERVICE, 10073692</td>
<td>FANS 1A installation/Installation of Unilink UU-801 connected to MICS 7120. Exchange of the DOVR and installation of 3 new switches.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): Flight Recorders and Data Link Recording CRF 0122-001 Non-rechargeable Lithium Battery Installations CRF 5128-001. The following paragraph(s) at a later amendment:</td>
<td>Master Document List DFS-DE-17-FS-0737 issue A- Airplane Flight manual Supplement FS0-068- MML Supplement DFS- DE-06-FS-A1 issue Originator later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/07/2020</td>
<td>Active</td>
<td>10073692</td>
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<tr>
<td>DASSAULT FALCON SERVICE, 10073693</td>
<td>DASSAULT FALCON SERVICE, 10073693</td>
<td>DASSAULT FALCON SERVICE, 10073693</td>
<td>Honeywell Satcom MICS 7120 Installation/Installation of Touch Panel 8000 Satcom and PACX Magnatar CT30 are removed and the MICS 7120 satcom system is installed.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): Security protection of Aircraft systems and networks CRI F-5128-001. The following paragraph(s) at a later amendment:</td>
<td>Master Document List DFS-DE-17-FS-0738 issue A- MML Supplement DFS-DE-06-FS-A1 issue Originator later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/07/2020</td>
<td>Active</td>
<td>10073693</td>
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<tr>
<td>DASSAULT FALCON SERVICE, 10073694</td>
<td>DASSAULT FALCON SERVICE, 10073694</td>
<td>DASSAULT FALCON SERVICE, 10073694</td>
<td>ADS-B out (DO-268B) installation/Transporters are upgraded. Three annunciator switches added transom panel and two BA-440 and associated relays installed in baggage compartments provide ADS-B out.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment:</td>
<td>Master Document List DFS-DE-17-FS-0736 issue A- Airplane Flight manual Supplement FS0-068- MML Supplement DFS- DE-06-FS-A1 issue Originator later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/07/2020</td>
<td>Active</td>
<td>10073694</td>
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<td>AIRWOLF FILTER CORP, 10073698</td>
<td>AIRWOLF FILTER CORP, 10073698</td>
<td>AIRWOLF FILTER CORP, 10073698</td>
<td>Installation of Airwolf Model AFC-500 Full Flow Engine Oil Filter P/N AFC-500</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SH10695N. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original...</td>
<td>Installation Drawing List AFOC-5001, dated 5 December 1992 and Installation Instructions No. AFC-K004-B, dated 5 December 1992.</td>
<td>Installation is limited to Schweizer 269A, B, C Helicopters only, Oering engine models O-360-B1A, HO-D-360-B1B, HO-D-360-A1A, HO-D-360-B1A, HO-D-360-B1B, HO-D-360-D1A, O-360-C2D. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other...</td>
<td>03/07/2020</td>
<td>Active</td>
<td>10073698</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>1007369</td>
<td>FOKKER TECHNIEK B.V.</td>
<td>REV. 1</td>
<td></td>
<td>A319-153N</td>
<td>A319 NEO Corporate Jet VIP conversion - Project ARCTIC</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The paragraph(s) at a later amendment: CS25 Amendment 21 for all project impacted paragraphs (see A319-ARCTIC- CRR-0001 issue 1) except 25.853 (f), 25.144(c), 25.13...</td>
<td>A319-ARCTIC-CRR-0001 Issue 1 dated 2 July 2020 A319-ARCTIC-AMMEL-S-0001 Issue 2 dated 23 June 2020 A319-ARCTIC-AMF-S-0001 Issue 2 dated 5 June 2020 A319-ARCTIC-CCOM-S-0001 Issue 2 dated 5 June 2020</td>
<td>06/07/2020</td>
<td>Active</td>
<td>1007369</td>
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<tr>
<td>10073700</td>
<td>FOKKER TECHNIEK B.V.</td>
<td></td>
<td>A319-153N</td>
<td>Installation of a large antenna for GOOG 2KU satellite communications system</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Equivalent Safety Findings(a), (b) R4-3601-006 Vibration and buffeting compliance criteria for large external antenna installations. Refer to the Appendix. The requirements...</td>
<td>A319-ARCTIC-CRR-0002 Issue 1 dated 02 July 2020 A319-ARCTIC-AMF-S-0002 Issue 1 dated 30 March 2020 A319-ARCTIC-AMF-S-0002 Issue 1 dated 26 March 2020</td>
<td>03/07/2020</td>
<td>Active</td>
<td>10073700</td>
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<td>10073703</td>
<td>AIRWOLF FILTER CORP</td>
<td>FAA STC SN00568NY</td>
<td>F-28F, 280F, 280FX, F-28A, 280, F-28C, 280C</td>
<td>Installation of remote mounted engine oil filter, model AFC-500 or AFC-800, and kit AFC K014</td>
<td>The Certification Basis for the original product remains applicable to the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-25 Amendment 4: 1301.1, 1303.1, 1303.3-25 Amendment 1...</td>
<td>Airwolf Installation Instructions No. AFC-K014-I-II, dated 30 April/1996 Airwolf Maintenance Instructions AFC-K014-I-III, dated 30 April 1996.</td>
<td>06/07/2020</td>
<td>Active</td>
<td>10073703</td>
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<tr>
<td>10073713</td>
<td>AVONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td></td>
<td>650</td>
<td>Installation of dual Garmin Transponder GTX 3000 with Position SourceGPS 3000</td>
<td>The Certification Basis for the original product remains applicable to the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-25 Amendment 4: 1301.1, 1303.1, 1303.3-25 Amendment 1...</td>
<td>- Flight Manual Supplement ASR-2020-004-FMS-05-00, Am 50 Instructions for Continued Airworthiness ASR-2020-004-IF2-10-00, Am 00-00 Instructions for Installation ASR-2020-004- IF2-10-00, Am 0000-00 Instructions for Installation ASR-2020-004- IF2-10-00, Am 0000-00</td>
<td>06/07/2020</td>
<td>Active</td>
<td>10073713</td>
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<td>10073714</td>
<td>AIRWOLF FILTER CORP</td>
<td>SH1052NE. DATED 20.12.17.</td>
<td>R22</td>
<td>Installation of Airwolf Model AFC-500 or AFC-800 full flow engine oil filter and, Kit No. AFC-K003</td>
<td>The Certification Basis for the original product remains applicable to the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-25 Amendment 1...</td>
<td>Installation Drawing List Report AFC-R001-(x), dated 27 October 1992</td>
<td>22/07/2020</td>
<td>Active</td>
<td>10073714</td>
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<tr>
<td>10073718</td>
<td>AIRWOLF FILTER CORP</td>
<td>FAA STC SN00342NY</td>
<td>R44</td>
<td>Installation of Airwolf remote mounted engine oil filter kit AFC-K011</td>
<td>The Certification Basis for the original product remains applicable to the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-25 Amendment 1...</td>
<td>Installation Drawing List Report AFC-R003, dated 24 May 1995 and Installation Instruction AFC-K011-I, dated 24 May 1995.</td>
<td>07/07/2020</td>
<td>Active</td>
<td>10073718</td>
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<td>STC Holder</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>JANCO AMERICA, INC.</td>
<td>FFA STC ST0675SE</td>
<td>787-9 787-10</td>
<td>Installation of Jamco America “Venture” BC Seat Furniture Modification/Service Bulletin on Boeing Model 787-9 and 787-10 aircraft</td>
<td>The Certification Basis for the original product remains applicable to all aircraft with the same equipment configuration as the original product, unless specified otherwise.</td>
<td>Jamco America, Inc. Master Drawing List (MDL) NO583-D10 Revision D, dated 02 May 2020, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/07/2020</td>
<td>Active</td>
<td>10073728</td>
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<tr>
<td>NORTWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>FFA STC ST0533SE</td>
<td>777-300ER</td>
<td>Economy Class Cabin refresh to include reseating the seating configuration from 10 abreast seats, rework of the passenger service unit, installation of new class dividers, installation of new flight entertainment, refresh of all partitions, monumentation/partition/voidage refresh and installation of...</td>
<td>The Certification Basis for the original product remains applicable to all aircraft with the same equipment configuration as the original product, unless specified otherwise.</td>
<td>Northwest Aerospace Technologies, Inc. (NAT) Master Data List (MDL) NO583-D10 Revision D, dated 02 April 2018, or later Federal Aviation Administration (FAA) approved revision.Instructions for Continued Airworthiness (ICA) NO583-D70, Revision A, dated 6 April 2018, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/06/2020</td>
<td>Active</td>
<td>10073733</td>
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<tr>
<td>DASSAULT FALCON SERVICE</td>
<td>FFA STC SAG403CH</td>
<td>SR22T SR22 SR20</td>
<td>Installation of an Air Conditioning System, Kelly Aerospacekit KATS-14-003.</td>
<td>The Certification Basis for the original product remains applicable to all aircraft with the same equipment configuration as the original product, unless specified otherwise.</td>
<td>- DFS-DE-17-7X-7152 Issue 09 April 2015 or later revisions of the above listed document(s) approved and accepted by the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/07/2020</td>
<td>Active</td>
<td>10073737</td>
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<td>KELLY AEROSPACE THERMAL SYSTEMS, L.</td>
<td>FFA STC SAG403CH</td>
<td>737-7000</td>
<td>The Certification Basis for the original product remains applicable to all aircraft with the same equipment configuration as the original product, unless specified otherwise.</td>
<td>Installation Instructions as listed on NC-14-021, Revision D, dated 24 September 2015, or later FAA approved revision FAA approved Airplane Flight Manual Supplement, Doc. No. NC-14-021, Revision A, approved 3 December 2015(1).</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/07/2020</td>
<td>Active</td>
<td>10073747</td>
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<tr>
<td>2 EXCEL DESIGN LIMITED</td>
<td>CAA UK RESPONSIBILIT</td>
<td>B200</td>
<td>Antenna Installations, Relocation &amp; Removals – King Air B200</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>- Installed in accordance with documents: Installation Instructions/Document number 2XL168 LEF30A (1), Issue 1, dated 18th June 2020(1). - Manned is in accordance with documents: Instructions for Continued Airworthiness document number 2XL168 LEF30A(1), Issue 1, dated 18th June 2020(1).</td>
<td>Aircraft build standard is a Beechcraft King Air 200 with Pratt &amp; Whitney Canada PT6A-42 engines with following four pre-requisite STCs/EASA STC 10023734 – Installation of Garmin GTX 360/7XX and GTX335/330/330X/Series/Flight Management System/EASA STC 10059881 – Provisions installation of Garmin GPS500XMS Antennae/ EASA STC...</td>
<td>07/07/2020</td>
<td>Active</td>
<td>10073751</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>BODGI AERONAUTICS S.R.L.</td>
<td>External Heated Mirror Installation (ref 109P00600).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any...</td>
<td>Master Document 109P00600-M01-01, dated 07 July 2020. Rotorcraft Flight Manual Supplement, 109P00600-FMS-01, Issue 01, dated 07 July 2020 or later revisions of the above listed document(s) approved/applete under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/07/2020</td>
<td>Active</td>
<td>10073760</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>Installation of Garmin GTN650H (Option A/IFR) and GNS430W (Option B/VFR).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any...</td>
<td>Document List 2112809 DOC, Issue 01, dated 10 July 2020. Rotorcraft Flight Manual Supplement for GTN650, 2112809-AN14, Issue 05, dated 10 July 2020. Rotorcraft Flight Manual Supplement for GNS430, 2112809-AMB, Issue 01, dated 02 July 2020. SD compliance 2112809-05, Issue 3, dated 14 January 2019. Refer to APP...</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/07/2020</td>
<td>Active</td>
<td>10073765</td>
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<td>SABENA TECHNICS BGC</td>
<td>OAE installation and Ka-band &amp; Wi-Fi systems activation on AirbusA340-600CJ</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements (Equivalent Safety Findings): CR B-01-1006641 Vibration/buffeting compliance criteria for large external antenna systems (in the following paragraph(s): at a later amendment...</td>
<td>- Master Data List, Aerocadoil document No. 0850-01-A-MLD-R00, dated 15 July 2020. Instructions for Continued Airworthiness, Aerocadoil document No. 0850-01-A-MLD-R01, dated 01 July 2020. Airplane Flight Manual Supplement, Aerocadoil document No. 0850-01-A-AMFIS-R00 (+/-) dated 03 June 2020 or later.</td>
<td>Limitations as per §3 of Instructions for Continued Airworthiness documents and §3 of Airplane Flight Manual Supplement documents referenced above Accomplishment of this STC requires the previous or simultaneous accomplishment of the following modifications: Mechanical and Electrical provisions installa...</td>
<td>13/07/2020</td>
<td>Active</td>
<td>10073770</td>
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<td>DUNCAN AVIATION, INC.</td>
<td>Installation of a R/H FWD Cabin Pocket Door</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval, except where amended by additional or later amendments if indicated on FAA STC 2071596W. The requirements for environmental protection and the associated certified noise and/or emissions levels of the...</td>
<td>Duncan Aviation, Inc. STC Master Document List, Document No. EDOC-0000017465, Revision A, dated 5 September 2014, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/07/2020</td>
<td>Active</td>
<td>10073778</td>
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<td>STANDARD AERO BUSINESS AVIATIONSERVIC ES, L.L.C.</td>
<td>Installation of an AlliedSignal Avionics, Dual GNS-XLS Flight Management System, with GPS as a primary means of navigation</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval, except where amended by additional or later amendments if indicated on FAA STC 2071596W. The requirements for environmental protection and the associated certified noise and/or emissions levels of the...</td>
<td>Garrett Aviation Services, Document List DL045, dated 16 October 1996, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/07/2020</td>
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<td>10073783</td>
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<td>Rev.</td>
<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>ARROW FILTER CORP</td>
<td>FAA STC</td>
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<td>Installation of Oil Filter Adapter and Oil Filter</td>
<td>FAA STC 12009-01, dated 13 April 2019</td>
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<td>14/07/2020</td>
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<td>GARMIN</td>
<td>FAA STC</td>
<td></td>
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<td></td>
<td>Modification of the landing gear warning system on aircrafts equipped with Garrett TFE731-1 engines and equipped with Collins ADC-85 Air Data Computers, to inhibit landing gear warnings above 160 knots.</td>
<td>Certification basis for the original product remains applicable to the certification approval, except where amended by additional or later amendments. If indicated on FAA STC 101009-07. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged.</td>
<td></td>
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<td>16/07/2020</td>
<td>Active</td>
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<td>INNOTEC AVIATIONA DIVISION OF IMP GROUP LTD</td>
<td>FAA STC</td>
<td></td>
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<td>Installation of G1000NXi Glass Cockpit Upgrade</td>
<td>Master Drawing List, Revision D, dated 09 October 2019, or later FAA approved revision. Aeronautical Accessories Instructions for Continued Airworthiness, Report No. AA-17597, Revision A, dated 19 March 2019, or later FAA approved revision. DECA Aviation Engineering Ltd. document 190-010-005-01, dated 01 June 2019.</td>
<td></td>
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<td>16/07/2020</td>
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<td>INNOTEC AVIATIONA DIVISION OF IMP GROUP LTD</td>
<td>FAA STC</td>
<td></td>
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<td></td>
<td>Installation of Garmin G3X Touch Electronic Flight Instrument Upgrade</td>
<td>Installation of the version of the GMA 245R with P/N: 010-01364-11(010-01471-12) is not approved as part of this STC. Previous P/N of the GMA 245R are still authorized. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td></td>
<td></td>
<td>23/04/2021</td>
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### Notes
- EASA STC no. and FAA STC no. for each STC entry.
- Certification basis for each product.
- Limitations/Conditions for each STC.
- Date of Issue for each STC.
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<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10073840</td>
<td>ASMAC AEROSPACE (SWITZERLAND AG)</td>
<td></td>
<td>A319-133</td>
<td>Cabin Recomfiguration/The Interior Outfitting Project includes a reconfiguration of the previously installed forward/forward seating area to accommodate additional or alternative airworthiness requirements. The certification basis for the original product remains applicable to this certificate/approval without any impact on the noise database.</td>
<td>The certification basis for the original product remains applicable to this certificate/approval without any impact on the noise database.</td>
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<td>21/07/2020</td>
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<td>10073840</td>
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<tr>
<td>10073842</td>
<td>LUTTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Thermo Pads: The Thermo Pad is a layer compound that is installed under the non textile floor (NFT) to provide additional electrical heating. The Pad warms up the floor panels, when connected to electrical power.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval without any impact on the noise database.</td>
<td>ES, L.L.C. AVIATIONSERVICES BUSINESS STANDARD AERO</td>
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<td>None</td>
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<td>10073843</td>
<td>STANDARD AERO BUSINESS AVIATIONSERVICES ES, L.L.C.</td>
<td>FAA STC SAB726L-0</td>
<td>MYSTÈRE-FALCON 20 - FS, MYSTÈRE-FALCON 20 - ES, MYSTÈRE-FALCON 20 - DS, MYSTÈRE-FALCON 20 - CS</td>
<td>Installation of three computer emergency batteries (two optional installations).</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval without any impact on the noise database.</td>
<td>AAAI Signal Aerospace Company, General Aviation Services Division drawing list DL972, no revision, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair must be determined.</td>
<td>21/07/2020</td>
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<td>10073843</td>
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<tr>
<td>10073844</td>
<td>STANDARD AERO BUSINESS AVIATIONSERVICES ES, L.L.C.</td>
<td>FAA STC SAB946L-0</td>
<td>2. MYSTÈRE-FALCON 20 - FS, 2. MYSTÈRE-FALCON 20 - ES, 2. MYSTÈRE-FALCON 20 - CS, 1. FAN-JET FALCON SERIES G, 1. FAN-JET FALCON SERIES F, 1. FAN-JET FALCON SERIES E, 1. FAN-JET FALCON SERIES D, 1. FAN-JET FALCON SERIES C</td>
<td>Installation of cabin interior panels consisting of the headliner, window panels, and sidewalls.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval without any impact on the noise database.</td>
<td>Garrett General Aviation Services Division drawing list DL974, no revision, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair must be determined.</td>
<td>21/07/2020</td>
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<td>10073845</td>
<td>STANDARD AERO BUSINESS AVIATIONSERVICES ES, L.L.C.</td>
<td>FAA STC SAB966L-0</td>
<td>3. MYSTÈRE-FALCON 20 - FS, 3. MYSTÈRE-FALCON 20 - ES, 3. MYSTÈRE-FALCON 20 - CS, 2. MYSTÈRE-FALCON 200 - FS, 1. FAN-JET FALCON SERIES G, 1. FAN-JET FALCON SERIES F, 1. FAN-JET FALCON SERIES E, 1. FAN-JET FALCON SERIES D, 1. FAN-JET FALCON SERIES C, 1. FAN-JET FALCON SERIES A</td>
<td>Installation of a Master Caution/Warning Panel.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval without any impact on the noise database.</td>
<td>Garrett Aviation Services Division drawing list GS-9200-011, no revision, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair must be determined.</td>
<td>21/07/2020</td>
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<td>Status</td>
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### Installation of a Collins TCAS-94 Traffic and Collision Avoidance System/TCAS II Transponders

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any im...

**Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.**

**16/03/2023** Active 10073684

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**Installation of oil filter adapter and oil filter Engine compartment firewall for mounting of filter shall be of 0.02inch minimum thickness ASTM AS47 galvanized steel or equivalent (Ref CAR 3.5(34)), FAR 23.1191(h)). Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.**

**23/07/2020** Active 10073688

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**Installation of a Collins TCAS-94 Traffic and Collision Avoidance System/TCAS II Transponders**

The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any im...

**Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.**

**22/07/2020** Active 10073846

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**Installation of oil filter adapter and oil filter Engine compartment firewall for mounting of filter shall be of 0.02inch minimum thickness ASTM AS47 galvanized steel or equivalent (Ref CAR 3.5(34)), FAR 23.1191(h)). Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.**

**23/07/2020** Active 10073844

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**Installation of oil filter adapter and oil filter Engine compartment firewall for mounting of filter shall be of 0.02inch minimum thickness ASTM AS47 galvanized steel or equivalent (Ref CAR 3.5(34)), FAR 23.1191(h)). Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.**

**23/07/2020** Active 10073844

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**Installation of oil filter adapter and oil filter Engine compartment firewall for mounting of filter shall be of 0.02inch minimum thickness ASTM AS47 galvanized steel or equivalent (Ref CAR 3.5(34)), FAR 23.1191(h)). Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.**

**23/07/2020** Active 10073844

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**Installation of oil filter adapter and oil filter Engine compartment firewall for mounting of filter shall be of 0.02inch minimum thickness ASTM AS47 galvanized steel or equivalent (Ref CAR 3.5(34)), FAR 23.1191(h)). Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.**

**23/07/2020** Active 10073884
<table>
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<th>STC Holder</th>
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<th>Model</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>AIRWOLF FILTER CORP</td>
<td>FAA STC SE4494NM</td>
<td>GO-480-G078, GO-480-G136, GO-480-G026, GO-480-G016, GO-480-G176, GO-480-F446, GO-480-F8, GO-480-F1A8, GO-480-F2A8, GO-480-C2C6, GO-480-C2D6, GO-480-B, GO-480-B1A6, GO-480-B1D, GO-480-C1D6, GO-480-B1B, GO-480-B1C</td>
<td>Installation of oil filter adapter and oil filter</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SE4494NM. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product.</td>
<td>FAA approved Mr. David P. Evans Inc. Master Drawing List, Drawing No. 0001, dated 1 September 1984, Revision N/C, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/07/2020</td>
<td>Active</td>
<td>10073747</td>
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<td>ARMSTRONG AEROSPACE Corp</td>
<td>FAA STC ST02313CH</td>
<td>A320-214, A321-112, A319-114, A319-11S</td>
<td>Installation of satellite TV System provisions</td>
<td>The Certification Basis for the original product as amended whereby following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25 at amdt 0: 25.23(a)(b), 25.29(4a)(b), 25.291(a)(1), 25.303, 25.305(a)(b), 25.307(a), 25.36, 25.61 ...</td>
<td>Armstrong Aerospace Master Data List, Doc. No. DC104-3101-00, Revision A, dated 26 May 2015, or later FAA approved revision.</td>
<td>The Satellite TV system equipment for which these provisions are amended or not covered by this approval an additional approval is required for the installation of this equipment. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/01/2021</td>
<td>Active</td>
<td>10073801</td>
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<tr>
<td>LIEBELLU MONDE INC.</td>
<td>FAA STC ST02510CH</td>
<td>A320-214, A321-112, A319-114, A319-11S</td>
<td>Installation of exterior placards, graphics and signs</td>
<td>The Certification Basis for the original product as amended whereby following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 amdt 11: 25.1, 25.601, 25.603, 25.605, 25.609, 25.611, 25.1301, 25.1305, 25.1501, 25.1529. The requirements for install...</td>
<td>Master Drawing List MDL-Pa-15-006 Rev McNeal. Instructions for Continued Airworthiness No. ICA-PA-17-025 Revision 2. To carry out the installation of the exterior placards, graphics and signs.</td>
<td>Installation and servicing instructions as per ICA document. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>14/08/2023</td>
<td>Active</td>
<td>10073869</td>
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<tr>
<td>AVIATION FABRICATORS, Inc.</td>
<td>FAA STC ST02727LA</td>
<td>A321-211-232, A321-211-212-213</td>
<td>Installation of cargo loading systemRevision 1: Adding of models A321-211-212-213 to the applicability list. Revision 2: Adding of models A321-213 back to the applicability list.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environment control and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any in...</td>
<td>Aviation Fabricators, Inc., Master Drawing List. AF-268MDL, Rev (B), dated 18 June 2011. Or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU-LCA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/07/2020</td>
<td>Active</td>
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<tr>
<td>ANORA INTERNATIONAL LLC</td>
<td>FAA STC ST02727LA</td>
<td>A321-211-232, A321-211-212-213</td>
<td>Installation of cargo loading systemRevision 1: Adding of models A321-211-212-213 to the applicability list. Revision 2: Adding of models A321-213 back to the applicability list.</td>
<td>The Certification Basis (CB) for the original product as amended whereby following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25 at amdt 16: 25.29(a)(b), 25.625(a)(b), 25.1557(a), 25.305(a)(b), 25.351, 25.613(a)(b), 25.853, 25.855, 25.1301(a)(b).</td>
<td>FAA approved Anora International LLC Controlled Data List DL8550 Revision 0, dated 3 February 2020, or later FAA approved revision. Instructions for Continued Airworthiness, Report Number 10551, Revision A, or later FAA accepted revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product/EASA STC 10071994 (FAA STC ST04405NY) must be installed...</td>
<td>03/11/2022</td>
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</table>
Installation of Airwolf remote mounted engine oil filter at AFC-K017 on 15tn aircraft powered by Pratt & Whitney radial reciprocating engine series R-985, R-1340 and R-1830.

The Certification Basis for the original product remains applicable to this certificate/approval, except where amended indicated on FAA STC SA1805NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remains unchanged and remain applicable to this certificate/approval without any in...

Airwolf Instructions for Continued Airworthiness, AFC-K000-IQA, revision FF, dated 5 July 2012, or later FAA accepted revision. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product. 1. Airwolf Oil Filter Model AFC-400 (Champion) is eligible only with R-985 series engines. 2. Airwolf Oil Filter Model AFC-700 (Fram) is eligible on R-885, R-1340and R-1830 series engines. 3. Engine compartment firewall for mounting of oil filter shall be of 0.021inch minimum thickness ASTM A527 galvanized steel or equivalent.

The Certification Basis for the original product remains applicable to this certificate/approval, except where amended indicated on FAA STC SA1805NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remains unchanged and remain applicable to this certificate/approval without any in...

Seaton Engineering Corporation Master Drawing List as listed on AMLSA01861SE. Instructio ns for Continued Airworthiness (IC) Document Number9910-008, Revision D, dated 25 June 2013, or later FAA approved revision. FAA approved Aircraft Flight Manual Supplement as listed on AMLSA01861SE.

Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and models approved.

Cabin control console and electrical system installation

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment : CS.27 Annex IV 40.67.1, 27.21, 27.25, 27.27, 27.29, 27.30, 27.305, 27.307, 27.309, 27.317, 27.337, 27.341...

- 135P0010-FMS-01 Flight Manual Supplement, Issue 1. 135P0010-M-00 Master Document List, Issue 1 for later revisions of the above listed document(s) approved/accepted by the FAA.

- Installation of STC EASA-IA.M.R.5.0140.Air Film Camera System is acceptable. Safety issues identified by the user for installation of this STC are unchanged and remain applicable to this certificate/approval.

- Installation of STC EASA-IA.M.R.5.0140.Air Film Camera System is acceptable. Safety issues identified by the user for installation of this STC are unchanged and remain applicable to this certificate/approval.

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): ASA CRI F-04 HIRF Protection Electronic Equipment Issue 37th requirements for environmental protection and the associated certified noise and/or emissions levels...

MASTERA DATA LIST SA19-015-MLR Rev. 00, SA19-015-AMMS-ESR Rev. 00, SA19-015-AMMS-VIS Rev. 00, SA19-015-AMMS-FMS Rev. 00, SA19-015-AMMS-REV 00, SA19-015-AMMS-REV 008r later revisions of the above listed document(s) approved/accepted by the SEE STC system.

03/07/2020 Active 10073900

The Certification Basis for the original product remains applicable to this certificate/approval, except where amended indicated on FAA STC ST01282NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remains unchanged and remain applicable to this certificate/approval without any in...

SA19-015-AMMS-ESR REV. 00, SA19-015-AMMS-VIS Rev. 00, SA19-015-AMMS-FMS Rev. 00, SA19-015-AMMS-REV 008r later revisions of the above listed document(s) approved/accepted by the SEE STC system.

03/07/2020 Active 10073900
List of EASA Supplemental Type Certificates

certification basis

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<tr>
<td>10073943</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>A320-231N, A321-251NX</td>
<td>Head Injury Criterion compliance modifications</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS25. Amptl, 10.25.96(c),11.58, and 11.59, and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval. The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): Non-rechargeable Lithium Battery Installation(s).</td>
<td>Master Drawing List ST-CB-013 rev 1, Aircraft Documents List, FAA Aircraft Noise and Emissions and EASA. Installation of one of the cabin layouts defined by the following documents as a prerequisite: A320 Aircraft Document 320-25-4777 rev A. The installation of this change/repair must be determined by the airworthiness community beforehand.</td>
<td>09/03/2020</td>
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<tr>
<td>10073948</td>
<td>HELI AUTRIA GmbH</td>
<td>212</td>
<td>Installation of a dual GTN 750 COMNAV/GPS including a GTX 349 transponder and a GRAS550 Radar Altimeter</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): Non-rechargeable Lithium Battery Installation(s).</td>
<td>Master Drawing List ST-CB-013 rev 1, Aircraft Documents List, FAA Aircraft Noise and Emissions and EASA. Installation of one of the cabin layouts defined by the following documents as a prerequisite: A320 Aircraft Document 320-25-4777 rev A</td>
<td>09/04/2020</td>
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<tr>
<td>10073951</td>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>REV. 1 FAA ST10936C</td>
<td>Installation of a TIA Air Coffee MK-1 coffee maker (model 400-2153-01) Revision 1, Transfer of certificate to Safran Cabin Germany GmbH.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List ST-CB-0113, Aircraft Documents List, FAA Aircraft Noise and Emissions and EASA. The installation of one of the cabin layouts defined by the following documents as a prerequisite: A320 Aircraft Document 320-25-4777 rev A</td>
<td>09/10/2020</td>
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<td>10073968</td>
<td>AIR PROJET</td>
<td>REV. 1 CAP108 CAP10</td>
<td>Installation of Ruder Hand control/Revision 1: Modification of the braking system</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval with no impact...</td>
<td>- Master Drawing List ST-CB-041 rev 1, Aircraft Documents List, FAA Aircraft Noise and Emissions and EASA. The installation of one of the cabin layouts defined by the following documents as a prerequisite: A320 Aircraft Document 320-25-4777 rev A</td>
<td>10/04/2022</td>
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<td>10073970</td>
<td>DRIF STIFTUNG LUFTRETTUNGEN EMENHUTZE AG</td>
<td>MBB-BK117 D-2 MBB-BK117 C-2</td>
<td>Installation of Multifunction EMS Groundplate.</td>
<td>The Certification Basis for the original product as amended the following paragraph(s) at a later amendment: CS-29 Amend. 42; 29.21, 29.25, 29.27, 29.29, 29.251, 29.303, 29.306. 29.307, 29.337, 29.651, 29.601, 29.603, 29.605, 29.607...</td>
<td>Certification Program Document No. CP-1117-25-50-00001 Rev. 01. Master Document: Document No. MD-1117-25-50-00049 Rev. 03. Installation manual, Document No. SA-1117-25-50-00019 Rev. 00. Instructions for Continued Airworthiness, Document No. CA-1117-25-50-00050 Rev. 00. Flight Manual Supplement, Document...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10073971</td>
<td>AIRWOLF FILTER CORP</td>
<td>FAA STC SA0453NY SEE FAA AMI</td>
<td>Installation of Airwolf remote mounted engine oil filter AFC-K010 onsingle and multi-engine fixed wing light aircraft and rotary wingcraft incorporating Franklin reciprocating engines of 450 horsepower less. Oil filter kit and installation instructions are specified in the attached Approved Mode...</td>
<td>The Certification Basis for the original product remains applicable if the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA0453NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remains applicable to this certificate/approval without any impact.</td>
<td>FAA STC SA0453NY From FAA STC AFC-K000-ICA, Revision IR, dated 5 July 2012, or later FAA accepted revision.</td>
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<tr>
<td>10073973</td>
<td>JET AVIATION AG</td>
<td>GIV-X</td>
<td>Oxygen Overboard Discharge Indicator.This STC covers the installation of a medical oxygen overboard discharge indicator installed on the fuselage between F3 193 and 257.5 and between stringers S19 and S20 LHS. There are no oxygen lines or equipment installed as part of this STC.</td>
<td>The Certification Basis for the original product remains applicable if the certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>AFM Supplement: JBSL-M-01456AFM-01 Issue 00. Instructions for Continued Airworthiness. JBSL-M-01456AFM-01 Issue 00. Later amendments of the above listed document(s) approved/accepted/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10073974</td>
<td>JET AVIATION AG</td>
<td>GII (G-1156A)</td>
<td>GPS Antenna Structural Installation</td>
<td>The Certification Basis for the original product as amended the following paragraph(s) at a later amendment: the following paragraphs of CS 25 at amendment 24, CS 25.23, CS 25.25, CS 25.261, CS25.301, CS 25.302, CS 25.303, CS 25.305, CS 25.30...</td>
<td>Master Data List, JBSL-M-03513MDL-001, Issue BB1 Drawing List (Mechanical). JBSL-M-03513MDL-001, Issue AE1 Engineering Order. JBSL-M-03513MDL-001, Issue Approval Instructions for Continued Airworthiness. JBSL-M-03513MDL-001, Issue A04 later revisions on file. Approved/accepted/accepted.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10073960</td>
<td>AEROSPACE DESIGN &amp; COMPLIANCE LLC</td>
<td>FAA STC ST04187NY 400, 400A</td>
<td>Installation of Encode Engine Data Acquisition Unit &amp; IMode DIGI Data Recorder on Textron Aviation Aircraft</td>
<td>The Certification Basis for the original product remains applicable if the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST04187NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Aerospace Design &amp; Compliance LLC Master Data List BE40-MDL-0139. Revision 1, dated 12 March 2019, original FAA approved/review (2) Master Data List BE40-MDL-0139. Revision 1, dated 20 November 2019, or later FAA accepted revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10073983</td>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>FAA STC ST04187NY 360-1751</td>
<td>Installation of Cabin Automatic Coffee Brewing System: Revision 1: Transfer of certificate to Safiran Cabin Germany GmbH</td>
<td>The Certification Basis for the original product remains applicable if the certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>TIA Master Drawing List 360-1751 Rev. 1, dated 18 August 2005.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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### List of EASA Supplemental Type Certificates

**EASA STC Holders**

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<th>STC Holders</th>
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<tr>
<td>STC Holder</td>
<td>REV. 1</td>
<td>A320-3231-3241-343</td>
<td>Additive Manufacturing Replacement Parts</td>
<td>This project is to replace existing Housing Assembly (Part number 2528R4155-202-00) on A330/P2 with a part designed and manufactured using Additive Manufacturing techniques</td>
<td>Revision 01: Scope extension for a part designed and manufactured using FLM technique.</td>
<td>03/07/2024</td>
<td>Active</td>
<td>10073985</td>
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<tr>
<td>PMV ENGINEERING</td>
<td>REV. 2</td>
<td>ATR 72-212 A</td>
<td>Medical Evacuation Operation for Air Taxi ifrev. 2: Typo Correction</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): D-0160073602 Medical evacuation configuration.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels are not changed and remain applicable to this certificate/approval without any impact.</td>
<td>04/06/2020</td>
<td>Active</td>
<td>10074002</td>
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<tr>
<td>WAYPOINT AEROTEC LIMITED</td>
<td>REV. 2</td>
<td>GV</td>
<td>Suppression of the EGWWS Aural Warning</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.26(b), 25.601, 25.603, 25.831, 25.853.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels are not changed and remain applicable to this certificate/approval without any impact.</td>
<td>11/08/2020</td>
<td>Active</td>
<td>10074003</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 2</td>
<td>GV</td>
<td>Installation of Honeywell CD-380 Control Display Units (CDU) on Gulfstream Model GV aircraft in accordance with Gulfstream Aerospace Corporation Index List CE51-1188M006, Revision B dated July 9, 2016.</td>
<td>Prior to installation of this change/repair it must be determined that the technical, airworthiness, and environmental requirements for the original product as amended by the listed document(s) approved/accepted under the EASA system.</td>
<td>26/08/2020</td>
<td>Active</td>
<td>10074010</td>
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<tr>
<td>SAFRING CABIN GERMANY Gmbh</td>
<td>REV. 1</td>
<td>FFA ST02856NY</td>
<td>Installation of a TIA Model 1547 Convection Oven, part number 40-0226-01/0 Revision 1: Transferee of certificate to Safran Cabin Germany Gmbh</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.305, 25.361 and CS 25.601, -</td>
<td>Overall envelope, clearances as per document ST-OV-11. Prior installation of this change/repair it must be determined that the technical, airworthiness, and environmental requirements for the original product as amended by the listed document(s) approved/accepted under the EASA system.</td>
<td>07/12/2020</td>
<td>Active</td>
<td>10074016</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>GV</td>
<td>Installation of Honeywell CD-380 Control Display Units (CDU) on Gulfstream Model GV aircraft in accordance with Gulfstream Aerospace Corporation Index List CE41 1188M008, Revision A, dated March 31, 2016.</td>
<td>Prior to installation of this change/repair it must be determined that the technical, airworthiness, and environmental requirements for the original product as amended by the listed document(s) approved/accepted under the EASA system.</td>
<td>24/06/2020</td>
<td>Active</td>
<td>10074017</td>
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For more information, visit [easa.europa.eu](http://easa.europa.eu).
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<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
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<tr>
<td>10074019</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 1</td>
<td>ST003580</td>
<td>A319-131/-132/-133 A319-111/-112/-113/-114/-115</td>
<td>Structure reinforcement for Ka-band antenna installation on Frame 53 and Airbus A319Rev. 1: - Extend the STC to cover a various smaller sizes of antennas installed in pre-mod in FR58. - Remove the limitations associated to the existing &quot;BBF&quot; antenna installed in FR58.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: CS 25.1.amd1.4 - CS 25.23.amd1.4 - CS 25.29.amd1.4 - CS 25.301.amd4.CS 25.303.amd4.CS 25.305.amd4.CS 25.306.amd4.CS...</td>
<td>- MDL 0910-02-A-MDL-R01 dated 04 February 2022: ICA 087-02-A-ICA-R01, dated 03 August 2020; later revisions of the above listed document(s) approved/accepeted under the EASA system.</td>
<td>10/02/2022</td>
<td>Active</td>
<td>10774107</td>
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<td>10074020</td>
<td>AERO TWIN, Inc.</td>
<td>FAA STC SA02391AK</td>
<td>KODIAK 100</td>
<td>A320-214</td>
<td>Installation of a Nose Gear Gravel Deflector, PN NGDQ-100</td>
<td>The Certification Basis for the original product remains applicable tois/d certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA02391AK. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are un...</td>
<td>Aero Twin, Inc. Installation Instructions, Document No. NGDQ-IM, Original Issue, dated 10 October 2011. or later FAA approved revision.Instructions for Continued Airworthiness, Aero Twin, Inc. NGDQ-ICA, Original Issue, dated 10 October 2011, or later FAA approved revision.</td>
<td>10/02/2022</td>
<td>Active</td>
<td>10774103</td>
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<tr>
<td>10074021</td>
<td>AIRBUS S.A.S.</td>
<td>FAA STC SA02391AK</td>
<td>KODIAK 100</td>
<td>A320-214</td>
<td>STANDALONE VHF/HF SYSTEM INSTALLATION AMD 0095: Installation and activation of a VHF/HF System on AC32020.</td>
<td>The Certification Basis for the original product remains applicable to is/d certificate/approval, except where amended by additional or alternative airworthiness requirements.- Equivalent Safety Finding(s): CRI B-4301-004. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product...</td>
<td>- MAS A00055 Issue 5-AIS-AJA-00086 issue 04 Certification Summary later revisions of the above listed document(s) approved/accepeted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2022</td>
<td>Active</td>
<td>10774101</td>
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<tr>
<td>10074023</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 1</td>
<td>ST003580</td>
<td>A319-131/-132/-133 A319-111/-112/-113/-114/-115</td>
<td>OAE installation on Frame 53 and Ka-band System Activation on AirbusA319Rev. 1: Cover a various smaller sizes of L-Band(SBB) antenna installed in pre-mod in FR58. - Remove the limitations associated to the existing &quot;BBF&quot; antenna installed in FR58.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- Equivalent Safety Finding(s): CRI B-01-100588: &quot;Vibration buffering compliance criteria for large external antenna installation&quot; Issue 1. - The following par...</td>
<td>MDL 0910-02-A-MDL-R01, dated 02 February 2022: AFMS 0910-02-A AFMS-R01, dated 16 July 2020. ICA 0510-02-A-ICA-R02, dated 31 January 2022; later revisions of the above listed document(s) approved/accepeted under the EASA system.</td>
<td>Limitations as per §3 of Aircraft Flight Manual Supplement document referenced above. Accomplishment of this STC requires the previous accomplishment of the following modifications:-Mechanical and Electrical Provisions, Minor Change Approvals 0909-01-A MCA-R03 or later approved revisions: - Aircraft Visual...</td>
<td>10/02/2022</td>
<td>Active</td>
<td>10774102</td>
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<td>10074025</td>
<td>AERO TWIN, Inc.</td>
<td>FAA STC SA02391AK</td>
<td>KODIAK 100</td>
<td>Installation of main gear gravel deflectors, PN NGDQ-100, on the mainlanding gear of the Quest Kodiak 100.</td>
<td>The Certification Basis for the original product remains applicable to is/d certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA02391AK. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are un...</td>
<td>Master Data List: G00-100, Original Issue, dated 14 February 2009, or later FAA approved revisions.Aero Twin, Inc.'s Instructions for Continued Airworthiness, Document Number G00-100-ICA. Original Issue, dated 14 February 2009, or later FAA Accepted revisions.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2022</td>
<td>Active</td>
<td>10774109</td>
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<tr>
<td>10074027</td>
<td>PRO STAR AVIATION, L.L.C.</td>
<td>FAA STC ST003580</td>
<td>BD-100-1A10</td>
<td>Installation of a Honeywell GoDirect Router</td>
<td>The Certification Basis for the original product remains applicable to is/d certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST003580. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are un...</td>
<td>Pro Star Aviation Installation Data List 05673-LD-101, Revision B dated 31 July 2020., or later FAA approved revision.Pro Star Aviation Airplane Flight Manual Supplement 05673-FMS-101 revision B, dated 15 July 2020., or later FAA approved revision.Pro Star Aviation Instructions for Continued Airworthi...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/08/2020</td>
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<td>10774107</td>
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<td>STC Holder</td>
<td>FAA STC or GDL Doc.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>Date of Issue</td>
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<td>System Record</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>GVI (G650)</td>
<td>Installation of a TrueNorth Iridium Telephone/WLAN System in accordance with Gulfstream Aerospace Corporation Index ListGDL100000015, Revision&quot;&quot;, dated July 14, 2015</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any im...</td>
<td>Underwater Locator Beacon Battery Replacement/Replacement of Underwater Locator Beacon (ULB) battery of the Cockpit Voice Recorder (CVR) system. The new Lithium Type Battery has improved performance (PIN DK12090) for 90 days duration.</td>
<td>The change is defined through Revision A of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implement...</td>
<td>17/09/2020</td>
<td>Active</td>
<td>A0074042</td>
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<tr>
<td>AMERICAN AVIATION Inc.</td>
<td>SA02615E</td>
<td>Installation of ram air cowling</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA02615E. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product remain unchanged.</td>
<td>American Aviation, Inc. &lt; &gt;-&lt; &gt; Drawing List 012 TLG, dated 29 May 2019, or later FAA approved revision American AviationRam Air Cowling Instructions for Continued Airworthiness Report 12-006, Revision N/ C, dated 24 June 2019, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/06/2020</td>
<td>Active</td>
<td>A0074045</td>
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<tr>
<td>AERO TWIN, Inc.</td>
<td>SA02344AK</td>
<td>Manufacture and installation of nose gear tire scraper, P/NTSQ-100, on the nose landing gear of the Quest Kodiak 100</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA02344AK. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product remain unchanged.</td>
<td>Master Data List, TSQ-100, Original Issue, dated 17 March 2009, or later FAA approved revisions.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/08/2020</td>
<td>Active</td>
<td>A0074046</td>
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<td>STC Holder</td>
<td>STC Code</td>
<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>ELLIOTT AVIATION TECHNICAL PRODUCTS DEVE LPMENT, LLC</td>
<td>FAA STC ST0424CH</td>
<td>400A</td>
<td>Installation of Automatic Dependent Surveillance - Broadcast (ADS-B) Outtransponder</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any change.</td>
<td>1. Elliott Aviation Master Data List Doc. No. MCL-3671-560 Revision C (&gt;</td>
<td>=) dated 18 May 2017, or later FAA approved revision. 2. Flight Manual Supplement (&gt;</td>
<td>=) Elliott Aviation Doc. No. 737-650-651, Revision IR, approved 19 May 2017 C (&gt;</td>
<td>=) or later FAA approved revision.</td>
<td>02/02/2020</td>
<td>Active</td>
<td>10074072</td>
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<tr>
<td>CORPORATION LOGIC AIR Inc.</td>
<td>TCCA SA19-31 ISSUE 2</td>
<td>737-800</td>
<td>Installation of GNSS with LPV Approach / ADS-B Systems</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendement: CS-ACNS initial issue: Subpart D; Section 8. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product.</td>
<td>LT82009 Rev. A – Modification Data Summary, LT82001 Rev. NC – Aircraft Flight Manual Supplement, LT82003 Rev. NC – Aircraft Flight Manual Supplement</td>
<td>20/08/2020</td>
<td>Active</td>
<td>10074074</td>
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<tr>
<td>PILATUS BUSINESS AIRCRAFT, Ltd.</td>
<td>FAA STC SA1003De</td>
<td>PC-24</td>
<td>Avance L3 system installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA1003De. The requirements for environmental protection and the associated certified noise and/or emissions levels of the...</td>
<td>- Master Document List for Avance L3 in a PC-24, 24-1806-01-0000, Revision C, dated 22 June 2020, or later FAA approved revisions Flight Manual Supplement for installation of the Avance L3 System. Document No. 24-1806-65-0000, Revision C, dated 17 June 2020, or later FAA approved revision.</td>
<td>25/08/2020</td>
<td>Active</td>
<td>10074092</td>
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<tr>
<td>JET AVIATION AG REV. 3</td>
<td>A319-153N</td>
<td>VIP Cabinet Interior CompletionThis modification installs a VIP Cabin Interior into a green AirbusA319-153N. The aircraft will be outfitted to accommodate 2 Cabin Crew and 19Passengers and can be operated non-commercial (private) as well as commercial. Two aircraft configurations are possible to be in...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s) CRI F-4308 Security Protection of Aircraft Systems and Networks. - The following paragraph(s) at a later amendment: CS 25 amendment 22</td>
<td>JBSC-M-0106MDL-01 ISSUE L MASTER DATA LISTBASELINE CONFIGURATION JBSC-M-0106FM-01 ISSUE E AIRPLANE FLIGHT MANUAL SUPPLEMENT JBSC-M-0108BICA-01 ISSUE A INSTRUCTIONS FOR CONTINUATIONWORTHYNESS JBSC-M-0108BICA-01 ISSUE B MASTER MINIMUM EQUIPMENT LIST SUPPLEMENT JBSC-M-0108BICA-01 ISSUE E CERTIFICATION.</td>
<td>Non-Smoking Aircraft. Applicable to Baseline Configuration only. As incompatibility demonstration with the certification basis for O/EConsortiums CCD and FCD has been performed, this aircraft is not allowed to be registered in any EASA member state without demonstration of compliance to O/E CCD and FCD...</td>
<td>10/03/2023</td>
<td>Active</td>
<td>10074094</td>
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<td>STC Holder</td>
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<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>DOA21 LIMITED</td>
<td>REV. 1</td>
<td>A320-251N</td>
<td>A320-251N</td>
<td>Introduction of a VIP Cabin Interonner: Transfer of STC holder role to DOA21 Ltd.</td>
<td>The Type and OSD Certification Bases for the original product is amended by the following additional or alternative airworthiness requirements: The following paragraph(s) is/are added to a later amendment: CS25 Amendment 24.25.7823, 85625.1365.1 Appendix 8.5C25-1, 5C25-10, 5C25-20, 5C25-30, 5C25-40, 5C25-50.</td>
<td>FAA STC</td>
<td>This design change (221-2563-1683) installs external doubliers on the fuselage skin between frames C22-C29 and C33-C34 which obscure the GVI required by task 552165-01-01 of Airbus A380 Part 2 Revision 08, as amended by EASA AD 2020-00561.</td>
<td>24/10/2023</td>
<td>Active</td>
<td>10074112</td>
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<tr>
<td>ANORA INTERNATIONAL LLC</td>
<td>10074113-02</td>
<td>STC733L</td>
<td>A321-251/232</td>
<td>A321-211/212-213</td>
<td>Installation of lower cargo deck loading system</td>
<td>The Certification Basis for the original product remains applicable to the STC.</td>
<td>FAA STC</td>
<td>This STC is not compatible with airplanes with auxiliary fuel tanks located in the lower cargo compartment. Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and/or repair will/introduce no adverse effect upon the airworthiness of the product. This modification does not impact the existing electric...</td>
<td>26/08/2020</td>
<td>Active</td>
<td>10074113</td>
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<tr>
<td>VT DRB AVIATION CONSULTANTS</td>
<td>10074114-01</td>
<td>STC7127DS</td>
<td>A320-320/-322/-341/-342</td>
<td>Interior reconfiguration increasing passenger count from 260 to 297 passengers.</td>
<td>Certification Bases as per TCDS amended by CS25 amendment 25.7272(b)(c), 25.7911(b)(c), 25.7956(a), 25.807(a)(b), 25.807(c)(d), 25.853(a)(b)(c), 25.864(a), 25.866(a)(b).</td>
<td>FAA STC</td>
<td>Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and/or repair will/introduce no adverse effect upon the airworthiness of the product. This modification does not impact the existing electric...</td>
<td>08/12/2020</td>
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<td>10074114</td>
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<tr>
<td>ARMSTRONG AEROSPACE Inc. d/b/a ARMSTRONG ASTRONAUTICS</td>
<td>REV. 1</td>
<td>A320-214</td>
<td>STC0448CH-D</td>
<td>Installation of an Aircraft Cabin Interiors Modification</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) is/are added to a later amendment: CS25 Amendment 25.29(a)(b), 25.303, 25.568(b)(d), 25.562(a)(b)(c), 25.601, 25.603(a)(b)(c), 25.719(b)(c), 25.7911(b)(c), 25.7956(a), 25.807(a)(b), 25.807(c)(d), 25.853(a)(b)(c), 25.864(a), 25.866(a)(b).</td>
<td>FAA STC</td>
<td>Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and/or repair will/introduce no adverse effect upon the airworthiness of the product. This modification does not impact the existing electric...</td>
<td>20/10/2020</td>
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<td>VT DRB AVIATION CONSULTANTS</td>
<td>10074119-01</td>
<td>STC0445CH-D</td>
<td>A320-320/-322/-341/-342</td>
<td>Interior reconfiguration increasing passenger count from 260 to 297 passengers.</td>
<td>Certification Bases as per TCDS amended by CS25 amendment 25.7272(b)(c), 25.807(a)(b), 25.807(c)(d), 25.853(a)(b)(c), 25.864(a), 25.866(a)(b).</td>
<td>FAA STC</td>
<td>Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and/or repair will/introduce no adverse effect upon the airworthiness of the product. This modification does not impact the existing electric...</td>
<td>08/12/2020</td>
<td>Active</td>
<td>10074119</td>
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### List of EASA Supplemental Type Certificates

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<th>Description</th>
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<th>Date of Issue</th>
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<th>System Record</th>
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<td>10074125</td>
<td>Q.G.M. DESIGN GmbH</td>
<td>Installation of Ku Band Satellite System with a Gimbal Antenna and WFI/Inpiranotes</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Thes following paragraph(s) at a later amendment: C5-25 Amendment 11, 21, 23, 25, 27, 29, 301, 303, 305, 307, 351, 571, 581, 601, 603, 605, 607, 609, 611, 613, 619, 625, 77...</td>
<td>- Aircraft Flight Manual Supplement BB07- QCM23-007-M1-APM, Rev. 2- Instructions for Continued Airworthiness BB07- QCM23-007-M1-ICA. Rev. 0 for later revisions of the above listed document(s) approved/ accepted/under the EASA system.</td>
<td>28/02/2020</td>
<td>Active</td>
<td>10074125</td>
</tr>
<tr>
<td>10074139</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>Uninterruptible Power Supply (UPS).Rev.1 extends the applicability of this STC to A319-133.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition (v); F-112 &quot;Superacoustic systems and installation&quot; issue 2. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>- Supplemental Type Definition Document STD224-200/1 Issue/Rev 02/00, dated 2 Feb 2021- For A321: Instructions for Continued Airworthiness A-00-2020/ICA/Rev 03/00, dated 30 Aug 2020- For A319-133: Instructions for Continued Airworthiness A-00-2101/ICA/Rev 01/00, dated 11 Jan 2021or L...</td>
<td>09/02/2021</td>
<td>Active</td>
<td>10074139</td>
</tr>
<tr>
<td>10074157</td>
<td>AIRLIFT A/S</td>
<td>Installation of external LED lights kit PN EN-169-33-101-01</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Design Order DO-169-33-101 Rev. 1 dated 28.01.2019/ Certification Plan CR-169-33-101 Rev. 4 dated 18.12.2019/ Compliance Checklist CLL-169-33-101 Rev. 0 dated 11.08.2020/ Flight Manual Supplement AW169 with Last light FMS-169-33-101-01 Rev. 1/ Instruction for Continued Airworthiness ICA-169-33-101-01 Rev. 1...</td>
<td>01/09/2020</td>
<td>Active</td>
<td>10074157</td>
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<tr>
<td>10074182</td>
<td>AVB PMA, LLC</td>
<td>EASA validation of FAA Parts Manufacturer Approval (PMA) NoPQ38885W/Supplement No 12 Revision 1, date July 15 2020; for the following FAA-PMA (for critical and other assemblies). NP-8UN441-1A Torque Link Pick-Up Sleeve Assembly, NPR-8UN101-1 Torque Link Pick-Up Sleeve, NP-8UN439-1AS...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>AVB PMA, LLC, PMA No: P038885W/S, Supplement No 12, Revision 1, date July 13, 2020; AVB MRO LLC Parts Manufacturer Approval Drawing List Document No: AVB- DL191150/Revision A, date 10/15/2019; AVB MRO, LLC Airworthiness Limitations Supplements documents AVB- ALS1911501 for the NP-8UN439-1A Steering...</td>
<td>04/09/2020</td>
<td>Active</td>
<td>10074182</td>
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<tr>
<td>10074187</td>
<td>AVIONIK STRAUBING ENTWICKLINGS GMBH</td>
<td>Installation of Single/Dual Research Equipment Carrier(s)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Installation in accordance with AvionicStraubing Master DocumentLatinified ASR-2019-066-MLD-01/00 dated 31.07.2020 and Instructions for installation ref ASR-2019-066-IFI-10/00 dated 08.07.2020 Operate in accordance with AvionicStraubing Aircraft Flight Manual/Supplement ref ASR-2019-066-FMS-05/00 date...</td>
<td>04/09/2020</td>
<td>Active</td>
<td>10074187</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>10074189</td>
<td>AIRBORNE MAINTENANCE ENGINEERING &amp; SERVICES, Inc. (AMES)</td>
<td>FAA ST04319CH</td>
<td>767-200, 767-300, 757-200</td>
<td>Installation of Automatic Dependent Surveillance – Broadcast (ADS-B) Outfitting Aviation Communication and Surveillance Systems (ACSS) NKT-800 Mode-S Transponders paired with Esterline/CMC Electronics OMA-5024 GPS/Sensors listed on AML number ST04319CH. The validated STC (ST04319CH) fully complies</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS ACNS Subpart D, Section 4. The requirements for environmental protection and the associated certified noise/ emissions levels of the product are unchanged/remain applicable</td>
<td>EASA MDL for ADS-B OUT on B767, Doc Ref: 47228/MDL03 Rev I EASA MDL for ADS-B OUT on B767, Doc Ref: 47228/MDL03 Rev I EASA AFM Supplement for B737-200 Series Aircraft, Doc Ref: 47228/AFM04 Rev I EASA AFM Supplement for B767-200/300 Series Aircraft, Doc Ref: 47228/AFM03 Rev I</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product</td>
</tr>
<tr>
<td>10074190</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>REV. 1</td>
<td>CL-600-2B19 (RJ SERIES 100) CL-600-2B16 (604 VARIANT) CL-600-2B16 (601-3R VARIANT) CL-600-2B16 (601-3A VARIANT) CL-600-2A12 (601 VARIANT) CL-600-1A11 (600)</td>
<td>Installation and activation of WiFi system to provide passengers with the means to access WAN services on their personal devices (PEDs).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: 25.1, 25.1301, 25.1309, 25.1501, 25.1529, 25.1561, 25.23.1585 at Amendment 24 of CS-25. The requirements for environmental...</td>
<td>Master Document List, S4A document No. 604102-12-13, Edition 4, dated 30 September 2020 Flight Manual Supplement, S4A document No. 604102-12-28, Edition 2, dated 4 August 2020 For this reason, any</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
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<tr>
<td>10074210</td>
<td>ELBIT SYSTEMS - ELECTRO-OPTICS-ELOP</td>
<td>REV. 1</td>
<td>757-300</td>
<td>Installation and activation of WiFi system to provide passengers with the means to access WAN services on their personal devices (PEDs).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: Amendment of the above following CS-25 requirements is CS-25(Amendment 11: CS-25.23 (b) CS-25.23 (a1) (a2)...</td>
<td>- B757 SPS Footprint Compliance Check List, ref. doc. 3759-0000-00GC, 20th March 2015 - B757 SPS Footprint Compliance Plan, ref. doc. 3759-0000-00CN, 20th March 2015 - B757 SPS Footprint Instruction for Continued Airworthiness (MPD Supplement), ref. doc. 3759-0000-002 Rev. B - B757 SPS Footprint Airplan...</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
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<tr>
<td>10074211</td>
<td>SWISS AIR-AMBULANCE Ltd.</td>
<td>REV. 1</td>
<td>MBB-BK117 D-3 MBB-BK117 D-2</td>
<td>Introduction of the White Phosphor NVG PIN PENN ING 600Rev. 1 extends the scope of this STC to the model MBB- BK117 D-3</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise/ emissions levels of the product are unchanged/remain applicable to this certificate/approval without any...</td>
<td>For initial issue/Design Effectivity List R980-00143 Rev. B dated 06.04.2020 Certification Programme R980-00143 Rev. B dated 06.04.2020 Compliance Summary List R980-00143 Rev. B dated 02.09.2020 Declaration of Compliance R980-00143 Rev. B dated 02.09.2020 Flight Manual Supplement R980-0021 Rev. C dated 02.09.2020</td>
<td>Rotorcraft modified by this design change employs Night Vision Imaging System lighting, commonly named NVIS lighting, which has been certifiedto ensure the aircraft is compatible with Night Vision Goggles (NVG). For this reason, any deviation from the cockpit or cabin configurations specified in this de...</td>
</tr>
<tr>
<td>10074212</td>
<td>SWISS AIR-AMBULANCE Ltd.</td>
<td></td>
<td>AW109SP</td>
<td>Introduction of the White Phosphor NVG PIN PENN ING 600</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise/ emissions levels of the product are unchanged/remain applicable to this certificate/approval without any...</td>
<td>Design Effectivity List R980-00132 Rev. B dated 06.04.2020 Certification Programme R980-00132 Rev. B dated 06.04.2020 Compliance Summary List R980-00132 Rev. B dated 02.09.2020 Declaration of Compliance R980-00132 Rev. B dated 02.09.2020 Flight Manual Supplement R980-00180 Rev. C dated 02.09.2020</td>
<td>Rotorcraft modified by this design change employs Night Vision Imaging System lighting, commonly named NVIS lighting, which has been certifiedto ensure the aircraft is compatible with Night Vision Goggles (NVG). For this reason, any deviation from the cockpit or cabin configurations specified in this de...</td>
</tr>
<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10074213</td>
<td>ELLIOTT AVIATION TECHNICAL PRODUCTS DEVE LPMENT, LLC</td>
<td>560XL</td>
<td>Installation of Mid-Continent Instruments MD-302 Stabilty Instrument on Textron Aviation Inc. 560XL aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact...</td>
<td>Elliott Aviation Master Data List MDL-4207-468, Revision E, dated 8 June 2020, or later FAA approved revision Airplane Flight Manual Supplement, Elliott Aviation Document No.FMS-4207-468, Revision A, FAA Approved 1 July 2020, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/06/2020</td>
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<tr>
<td>10074217</td>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>DHC-6 SERIES 400</td>
<td>Installation of Replacement Cabin Roof</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact...</td>
<td>Installation in accordance with Rocky Mountain Aircraft (RMA)(Modification Instruction E17111), Revision A, dated 05 June 2020. Maintenance in accordance with RMA Maintenance Manual Supplement E17113 1, Revision A, dated May 22, 2020, or later revisions of the above listed document(s) approved/ accepted.</td>
<td>Aircraft roof assembly must be pre-mod Viking Aircraft Ltd. Modification No. 611256 or equipped with E066397 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no...</td>
<td>08/06/2020</td>
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<tr>
<td>10074223</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td>B300, B300C, B200GT, B200, B200C, B200CCT, B200CCT-E, A100-1, A200, A200C, A200CT, 300, 300LW, 300, 300C, 300CT, 1900, 1900C, 1900D</td>
<td>RC2-652 transponder replacement with NKT-600 transponder for ADS-B Outcompliance in the above Textron Aviation Beechcraft Models (NON ProLine21 aircraft).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-ACNS Initial Issuance Section 3- Mode-S Elementary Surveillance; Section 3- Mode-S Enhanced Surveillance; Section 4- 1090MHz...</td>
<td>Installation Procedure, Report 2020-22-1 Revision A, dated 20 August 2020 Aircraft Flight Manual Supplement, Report 2020-22-6, Revision IR, dated 29 June 2020 Maintenance Manual Supplement Report 2020-20-10, Revision A, dated 28 August 2020, or later revisions of the above listed document(s) approved/ accepted.</td>
<td>Before the modification of the aircraft, it must be checked that the SBAS receiver is listed in the Table 1 of the Installation Instructions/Report 2020-22-1 Section 1 including the software version Prior to installation of this change/repair it must be determined that the interrelationship bet...</td>
<td>09/06/2020</td>
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<tr>
<td>10074225</td>
<td>AVIONITEC AG</td>
<td>340A</td>
<td>Avionics upgrade Installation of a T6S Top Antenna Garmin GA 58 in a pressurized Area(&gt;5. ) Installation of a L-3 WX-500 Stormscope System with new Antenna in non-pressurized area. Stormscope Information displayed and controlled by the Garmin GTN Display/Ec...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: § 23.887, 23.1308 and 23.1308 of CS 23 Amd 4 The requirements for environmental protection and the associated certified noise...</td>
<td>- Master Document List: Doc. No. 9469-017, Dated 31.08.2020- Aircraft Flight Manual Supplement: Doc. No. 9469-011, Date 28.01.2020- Instructions for continued airworthiness Doc. No. 9469-015, Date 29.03.2019 or later revisions of the abovementioned document(s) approved/ accepted under EASA system</td>
<td>Limited to Serial Number 340A-0948 (HB-LMN) Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/06/2020</td>
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<tr>
<td>10074232</td>
<td>ANORA INTERNATIONAL LLC</td>
<td>FAA STC ST02705LA</td>
<td>Installation of window plugs</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25 and AMD 00: 25.29, 25.303, 25.961, 25.601, 25.605, 25.607, 25.619, 25.620(a)(5), 25.843(a), 25.899(a)(1), 25.1525CS 25 am...</td>
<td>Ancra International LLC Controlled Data List DLST740, Revision D, dated 28 June 2019, or later FAA approved revision Instructions for Continued Airworthiness, Revision A9932, or later FAA accepted revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/01/2021</td>
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<td>Model List</td>
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<td>Limitations/Conditions</td>
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<tr>
<td>10074234</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>PA-31T (CHEYENNE/ CHEYENNE I) PA-31T2 (CHEYENNE II) PA-31T (CHEYENNE III) PA-31P (PRESSURIZED NAVAJO) PA-31P-350 (MOJAVE) PA-31-350 (CHEYATTAR) PA-31-300, PA-31-325 PA-31</td>
<td>Installation of an STC 01326 Digital Flight Control System (DFCS)</td>
<td>1. Master Data List No. ST-975-MDL-001, Revision 1, dated 23 June 2020 later FAA approved revision 2.</td>
<td>Prior to installation of this change/repair and any other previously installed change/repair will int...</td>
<td>15/09/2021</td>
<td>Active</td>
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<tr>
<td>10074237</td>
<td>STG AEROSPACE GERMANY GmbH</td>
<td>REV. 5 SEE EASA APPROVED MODEL LIST</td>
<td>The Certification Basis for the original product as amended by EASA STC 10042314</td>
<td>SystemsRev. 1: Transfer of STC holder to STG Aerospace Limited to STGAerospace GmbHRev. 2 incorporates the installation of a new SSUL-X variant of the STGAerospace s</td>
<td>SSUL-X Green Emitting PL insert shall not be mixed with SSUL-X Blue Emitting PL insert on the same aircraft. Installation of SSUL-X systems shall not be mixed with the aircraft cabin area.</td>
<td>02/06/2023</td>
<td>Active</td>
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<tr>
<td>10074240</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>31A FMS Upgrade with UNS-Lxw covering SBAS and RNP-1 Capability</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate's approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate's approval without any in...</td>
<td>According to QCM Design GmbH Document CR-L31X-QCM04-001, Rev. 1 it be operated in accordance with QCM Design GmbH AFM-S Document No.L31X-QCM04-001 AFM. Rev. 2To be installed as a QCM Design GmbH Document Engineering Bulletin EB-L31X-QCM04-01, Initial IssueTo be maintained as a QCM Design GmbH Doc.</td>
<td>This change is only for aircraft equipped as defined in chapter 1.1 V061M Engineering Bulletin EB-L31X-QCM04-001, Rev. 1 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will int...</td>
<td>11/06/2020</td>
<td>Active</td>
</tr>
<tr>
<td>10074241</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>REV. 1 SEE EASA APPROVED MODEL LIST</td>
<td>Replacement of Halon agent with alternate Halon-free Portable FireExtinguisher agent. This certification covers the replacement of the existing Portable FireExtinguisher agent with Halon-free Portable Fire Extinguisher agent at Halton B.V. Rev. 1 Remove Limitation for “Low Occupancy” aircraft and add...</td>
<td>The Change is defined with revision B of AMAC Aerospace Master Data ListDocument MDL-ALLAC01-2620. The installation has to be accomplished in accordance with revision B ofAMAC Engineering Order Document EO- ALLAC01-2620 later revisions of the above listed document(s) approved/accepted by the EASA...</td>
<td>Physical installation of the fire extinguisher is not covered by thiomodification and requires a concurrent approval. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will int...</td>
<td>14/01/2022</td>
<td>Active</td>
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<tr>
<td>10074267</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA09759DS PA-31T (CHEYENNE/ CHEYENNE I) PA-31T2 (CHEYENNE II) PA-31T (CHEYENNE III) PA-31P (PRESSURIZED NAVAJO) PA-31P-350 (MOJAVE) PA-31-350 (CHEYATTAR) PA-31-300, PA-31-325 PA-31</td>
<td>Installation of an S-TEC 01326 Digital Flight Control System (DFCS)</td>
<td>1. Master Data List No ST-975-MDL-001, Revision 1, dated 23 June 2020 later FAA approved revision 2. Airplane Flight Manual Supplement, ST-975-AMS-FMS-001, Rev. 1, dated 23 June 2020 later FAA approved revision 3. Instructions for Continued Airworthiness, ST-975-CA-001, Revision, dated 17 Jun...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will int...</td>
<td>15/09/2020</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC No(s)</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>INTESAT INFILIT LLC</td>
<td>REV. 3</td>
<td>FAA STC ST0241CH</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of an Aircraft Broadband Services (ABS) SystemRev1: Type and Certification Basis correctedRev2: Company Name changed from &quot;Gogo LLC&quot; to &quot;Intesat Infilit, LLC.&quot;</td>
<td>1. Installation instructions as listed on FAA AML Number ST0541CH:1. Airplane Flight Manual Supplement (AFMS) as listed on FAA AML Number ST0541CH.2. Instructions for Continued Airworthiness (ICA) as listed on FAA AML Number ST0541CH.</td>
<td>Equivalent Safety Findings have been made for the following: 25.813(c)(2) &quot;Emergency Exit Access&quot; (documented in TAD ELOS NexcomSP08431B-T-1). Prior to installation of this STC, prior to installation of the change/repair it must be determined that the interim relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/03/2021</td>
</tr>
<tr>
<td>ANI ALTITUDE UK LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>787-9</td>
<td>787-8</td>
<td>Installation of EL Alt Feature Panels C781AAAGA009-01</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval without any changes.</td>
<td>Prior to installation of this STC, prior to installation of the change/repair it must be determined that the interim relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/06/2020</td>
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<tr>
<td>STC HOLDING COMPANY, INC.</td>
<td>FAA STC SA7738SW</td>
<td>DHC-6 SERIES 100</td>
<td>DHC-6 SERIES 200 DHC-6 SERIES 100</td>
<td>Installation of a Freon Air Conditioning System</td>
<td>1. R. W. Martin Drawing List No. 89012, dated 18 September 1990, order FAA approved data 2. Flight Manual Supplement dated 13 March 1990, or later FAA approved.</td>
<td>The Model 100 must have de Havilland Modification Bulletin No. 61075 &quot;To Extend Rear Baggage Compart,&quot; installed for installation of thisSTC. Prior to installation of this change/repair it must be determined that the interim relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/09/2020</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC no.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>LB AIRCRAFT ENGINEERING LLC</td>
<td>REV. 1</td>
<td>FAA STC ST0447AT</td>
<td>737-300</td>
<td>FAA STC</td>
<td>1. LB Aircraft Engineering LLC. Master Data List LB-TRA375-MD1, RevisionB, dated 10 January 2020, or later FAA Approved Flight Manual Supplement. LB Aircraft Engineering document LB-TRA375-ATM, Rev IR, dated 13 March 2020, or later FAA Approved revisions.3. LB Aircraft Engineer...</td>
<td>Both FMCs must be loaded with operator specific Boeing provided OPCS/Cafoo software with GPS update recognition. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC approves the installation of eq...</td>
<td>28/01/2021</td>
</tr>
<tr>
<td>SWS CERTIFICATION SERVICES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A330-941</td>
<td>AZUL AIRBUS A330NEO RECORD CREW VOUCHER ECONOMY SEAT ACTIVATION</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval without any impact.</td>
<td>SWS Certification Services Ltd Master Configuration Scheme No 3183-140-101. FAA approved revisions.3. FAA approved revisions.2. FAA Approved Flight Manual Supplement. LB Aircraft Engineering document LB-TRA375-MD1, RevisionB, dated 13 March 2020, or later FAA Approved revisions.3. LB Aircraft Engineer...</td>
<td>Both FMCs must be loaded with operator specific Boeing provided OPCS/Cafoo software with GPS update recognition. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC approves the installation of eq...</td>
<td>21/06/2020</td>
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<tr>
<td>SAFRAN CABIN INC.</td>
<td>FAA STC ST01928A</td>
<td>A320-214, A321-25JN A319-112, A319-115</td>
<td>Installation of Forward LHRH Windows, Mid LHRH Class Divider, M/LHRH Windows, Upper Attach Provision and Interior Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval without any impact.</td>
<td>FAA approved Flight Manual Supplement. FAA approved revisions.3. FAA approved revisions.2. FAA Approved Flight Manual Supplement. LB Aircraft Engineering document LB-TRA375-MD1, RevisionB, dated 13 March 2020, or later FAA Approved revisions.3. LB Aircraft Engineer...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC approves the installation of eq...</td>
<td>09/11/2020</td>
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<tr>
<td>ESG ELEKTRONIKS ITM-UND LOGistik-GmbH</td>
<td>REV. 1</td>
<td>EC 155 B</td>
<td>EC 155 B</td>
<td>Installation of a mission management system (MMI) workstations and acameras system HD EOS/Revision 1 corrects references in Associated Technical Documentation and clarifies the Limitations/Conditions.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval without any impact.</td>
<td>- Flight Manual Supplement, Doc. No. 00044-DO-MD-FMS-00, Issue 01 dated 28 May 2020. - System Description, Doc. No. 00044-DO-MD-MD-00, Issue 03 dated 28 May 2020. - List of Compliance Documents, Doc. No. 00044-DO-MD- LCD-00, Issue 01 dated 13 August 2020. Instructions for Continued Airworthiness, Doc...</td>
<td>The activation of non-eye safe laser (Class 3, 3B and 4) in the aircraft must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC approves the installation of eq...</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAAA STC no.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10074336</td>
<td>WAS CONSULTANTS, INC. d/b/a TEL AIR INTERNATIONAL</td>
<td>REV. 2</td>
<td>FAA STC ST02736LA</td>
<td>SEE FAA AMI</td>
<td>Installation of modular lower deck Cargo Loading System (CLS). Rev. 1: Typos corrected; Rev. 2: A321 aircraft family included</td>
<td>The Certification Basis for the original product as amended by additional or alternative airworthiness requirements.</td>
<td>For A320, FAA approved Tair International Index List 03/01/10 Revision A, dated 28 August 2020. FAA approved Airplane Flight Manual Supplement 969101GS-FMS-2200 Revision n Original. Dated 31 July 2020. Weight &amp; Balance Control and Loading Manual, Report No. 255-51. Revision A, dated 5 June 2020</td>
</tr>
<tr>
<td>10074362</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST015426W-D</td>
<td>SEE FAA AMI</td>
<td>Modification of a Rockwell Collins. Inc. AFSD-3010 and AF3-3010E Adaptive Flight Display Installation in Part 25 Aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC ST015426W-D. The requirements for environmental protection and the associated certified noise and/or emissions levels of the aircraft.</td>
<td>Rockwell Collins&lt;()-()&gt;. Inc. Master Drawing List as listed on AML No. ST0715426W-D. Revision -,-, or later FAA approved revisions.</td>
<td>Prior installation and approval of a Rockwell Collins&lt;()-()&gt;. Inc. Master Drawing List as listed on AML No. ST0715426W-D. Revision -, or later FAA approved revisions.</td>
</tr>
<tr>
<td>10074396</td>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>REV. 1</td>
<td>777-300ER</td>
<td>Cabin Reconfiguration Rev. 1: Transfer of STC from SR Technics Switzerland to growing Engineering Partners GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the aircraft are unchanged and remain applicable to this certificate/ approval without any...</td>
<td>Master Document List S.S0AEZ.001918-AMF, Rev. 00Aircraft Flight Manual Supplement S.S0AEZ. 001918-AFM, Rev. 01Instruction for Continued Airworthiness S.OAEZ. 001918-ICA, Rev. 00Or later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>The installation of STC 10065375 is pre-configuration is a prerequisite. Compliance with the ODI Certification Basis of the product with this STC installed has not been shown. The product with this STC installed is therefore not eligible for registration in any EASA member state unless compliance with 1) ...</td>
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<tr>
<td>10074402</td>
<td>TEXAS AIR SERVICES, Inc.</td>
<td>FAA STC SA04078C</td>
<td>DHC-6 SERIES 400 DHC-6 SERIES 300 DHC-6 SERIES 200 DHC-6 SERIES 100</td>
<td>Installation of a replacement electrically driven by hydraulic pumpassembly</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA04078C. The requirements for environmental protection and the associated certified noise and/or emissions levels of the aircraft remain unchanged and are applicable to this certificate/ approval without any...</td>
<td>Texas Air Services, Inc. Master Drawing List No. AAP-001, Revision 4 dated 4 October 2016, or later FAA approved revision. Installation Instruction with Instructions for Continued Airworthiness, Document No. AAP-001, Revision 4 dated 4 October 2016, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is limited to...</td>
</tr>
<tr>
<td>10074403</td>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>TCCA C-LS230-0475</td>
<td>1900D</td>
<td>GTS 8000 TCAS II V7.1 and GTX 3000 ADS-B Out Upgrade. This modification installs a Garmin GTS 8000 TCAS II V7.1 system and a GTX 3000 transponder system with ADS-B Out capability, including combined TCAS/vertical speed indicators and structural provisions for TCAS and transponder antennas.</td>
<td>The Installation Basis for the original product as amended by additional or alternative airworthiness requirements.</td>
<td>Change defined in Master Data List CE480-900 Issue C, or later TCCA approved revision. Installation in accordance with Engineering Order CE480-401 Issue B, later TCCA approved revision. Installation in accordance with Maintenance Manual Supplement CE480-701 Issue C, or later TCCA approved revision.</td>
<td>This STC is only applicable to aircraft serial number UE-398. Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the...</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
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<td>10074404</td>
<td>FOKKER SERVICES B.V.</td>
<td></td>
<td>737-600ER 737-600 737-600 737-700</td>
<td></td>
<td></td>
<td>Activation of standalone GNSS sensor system on Boeing 737-MG Aircraft. The present STC concerns the activation of the dual standalone GNSS sensor system installed by means of minor mod ref C0916877 (CPR-0425). The stand-alone GNSS sensor system has no connection to any other aircraft system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any...</td>
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<tr>
<td>10074408</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING SERVICES ING. WALTER STARZACHER GbR</td>
<td></td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td></td>
<td></td>
<td>Replacement of Portable Fire Extinguishers using Halon agent with alternate Halon-free agent with existing Portable Fire Extinguishers agent (Halon Br-X).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25.1, CS25.25, CS25.301, CS25.303, CS25.305, CS25.307, CS25.561, CS25.601, CS25.603, CS25.605, CS25.607, CS25.609, CS25.61...</td>
</tr>
<tr>
<td>10074409</td>
<td>ARMSTRONG AEROSPACE Inc. d/b/a ARMSTRONG AEROSPACE</td>
<td></td>
<td>FAA STC ST02455CH</td>
<td></td>
<td></td>
<td>Installation of an In-Flight Entertainment System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25.20(a)(1), 25.303, 25.305(a)(5), 25.307(a)(5), 25.365(a)(2)(iv), 25.365(d)(2), 25.367(a)(2)(iv)(D), 25.401...</td>
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<tr>
<td>10074412</td>
<td>MECAG AIRAVATION GROUP S.p.A</td>
<td></td>
<td>H160-B (IN PROGRESS)</td>
<td>H160-B</td>
<td></td>
<td>“STYLENCE” cabin interior configurations family PIN 6A4H0000X001001 specified in doc. ref. 6A4H0000X001001 and 6A4H0000X001008. Each configuration incorporates installation of seats, cabinets (where applicable), carpet in passenger compartment together with existing trim panel modifications (i.e. painting...</td>
<td>The Type and OSD Certification Basis (CB) for the original product incorporates the applicable airworthiness requirements.</td>
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<tr>
<td>10074417</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td></td>
<td>FAA STC SA1544W-H-D</td>
<td>SEE FAA AML</td>
<td></td>
<td>Modification of a Rockwell Collins. Inc. AFD-3010 and AFD-3010E Adaptive Flight Display Installation in Part 23 Aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
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<tr>
<td>STC Holder</td>
<td>Active</td>
<td>GVH AEROSPACE</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>STC Holder</td>
<td>Active</td>
<td>A350-941</td>
<td>Cabin Integration/Cabin conversion from a previously configured standard airline cabin to governmental use.</td>
<td>The STC consists of the installation of the ADBS-OULT function on Airbus A340/Approved configurations are defined in the applicable Aircraft Flight Manual Supplement ref. AVI-02330- AFMS.</td>
<td>- Master Data Document AVI-02230-MDL-501-R01, dated 30 July 2020. Aircraft Flight Manual Supplement AVI-02230- AFMS-501- R05, dated 23 MAR 2020. Instructions for Continued Airworthiness AVI-02230- ICA-501-R01, dated 06 APR 2020. MMEL Supplement AVI-02230- MMELS-R05 rev R00, dated 23 MAR 2020 for later renewals.</td>
<td>STC limitations and approved configurations are defined in the applicable Aircraft Flight Manual Supplement ref. AVI-02230- AFMS. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair is not adverse effect upon the airworthiness of the aircraft.</td>
<td>02/10/2020</td>
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<tr>
<td>PMI ENGINEERING</td>
<td>Active</td>
<td>A340-313J/541J/642J/643 J/621J/212J/213J/311J/312J</td>
<td>The STC consists on the installation of the ADBS-OULT function on Airbus A340/Approved configurations are defined in the applicable Aircraft Flight Manual Supplement ref. AVI-02330- AFMS.</td>
<td>The STC consists of the installation of the ADBS-OULT function on Airbus A340/Approved configurations are defined in the applicable Aircraft Flight Manual Supplement ref. AVI-02330- AFMS.</td>
<td>- Master Data Document AVI-02230-MDL-501-R01, dated 30 July 2020. Aircraft Flight Manual Supplement AVI-02230- AFMS-501- R05, dated 23 MAR 2020. Instructions for Continued Airworthiness AVI-02230- ICA-501-R01, dated 06 APR 2020. MMEL Supplement AVI-02230- MMELS-R05 rev R00, dated 23 MAR 2020 for later renewals.</td>
<td>STC limitations and approved configurations are defined in the applicable Aircraft Flight Manual Supplement ref. AVI-02230- AFMS. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair is not adverse effect upon the airworthiness of the aircraft.</td>
<td>02/10/2020</td>
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<tr>
<td>FAN JET FALCON SERIES F</td>
<td>Active</td>
<td>FAN JET FALCON SERIES</td>
<td>Installation of 90-Day Underwater Locator Beacon. The existing 30-day CVR and FDR USBs are replaced with 90-day versions.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements - Special Condition (s): F-1128 Non-rechargeable lithium batteries and/ or battery installations. The following paragraph(s) at a later amendment: CS 25-404-1: CS25.1 (a), CS25.131(c), CS25.130(c)(2)(a)(b), CS25.123(c)(2)(d), CS25.123(a)(2), CS25.130(c)(1), CS25.130(c)(2).</td>
<td>Installation of 90-Day ULB doc. No. GVH-10367-03- MOD-01 Issue 1 dated 06 Sept. 2020 for later revisions of the above listed document(s) approved acceptable under the EASA system.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Subpart C, 21.109. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair introduction will not introduce an adverse effect upon the airworthiness of the aircraft.</td>
<td>02/10/2020</td>
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<tr>
<td>HAWKER 800XP, HAWKER 1000</td>
<td>Active</td>
<td>HAWKER 800XP, HAWKER 1000</td>
<td>Installation of 90-Day Underwater Locator Beacon. The existing 30-day CVR and FDR USBs are replaced with 90-day versions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im…</td>
<td>Installation of 90-Day ULB doc. No. GVH-10367-03- MOD-01 Issue 1 dated 06 Sept. 2020 for later revisions of the above listed document(s) approved acceptable under the EASA system.</td>
<td>In accordance with FAA/EASA TIP rev. 6, Article 7.11.3, the following statement should be written in the remarks block of the FAA Form 8130-3: &quot;Produced by the holder of the EASA STC number 10074450&quot;. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair introduction will not introduce an adverse effect upon the airworthiness of the aircraft.</td>
<td>02/10/2020</td>
</tr>
<tr>
<td>AV8 PMA, LLC</td>
<td>Active</td>
<td>AV8 PMA, LLC</td>
<td>Installation of 90-Day Underwater Locator Beacon. The existing 30-day CVR and FDR USBs are replaced with 90-day versions.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im…</td>
<td>Installation of 90-Day ULB doc. No. GVH-10367-03- MOD-01 Issue 1 dated 06 Sept. 2020 for later revisions of the above listed document(s) approved acceptable under the EASA system.</td>
<td>In accordance with FAA/EASA TIP rev. 6, Article 7.11.3, the following statement should be written in the remarks block of the FAA Form 8130-3: &quot;Produced by the holder of the EASA STC number 10074450&quot;. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair introduction will not introduce an adverse effect upon the airworthiness of the aircraft.</td>
<td>02/10/2020</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>Active</td>
<td>SAO9762DS</td>
<td>Installation of an S-TEC 01326 Digital Flight Control System (DFCS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SAO9762DS. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the on…</td>
<td>Master Data Document No. ST-973-MLD-0001, Revision 4, dated 30 June 2020, or later FAA approved revision Airplane Flight Manual Supplement, ST-973-AMS-000, Rev. 3, dated 30 June 2020 or later FAA approved revision Instructions for Continued Airworthiness, ST-973-ICA-0000, Revision - dated 4 June 2019.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair introduction will not introduce an adverse effect upon the airworthiness of the aircraft.</td>
<td>02/10/2020</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
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<td>Associated Technical Documentation</td>
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<td>10074474</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
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<td>MD-88</td>
<td>Underwater Locator Beacon Battery Replacement/Replacement of Underwater Locator Beacon (U LB) battery of the Cockpit Voice Recorder / Flight Data Recorder. The new Lithium Type Battery has improved performance (55AAR battery) for 90 days duration.</td>
<td>The Certification Basis for the original product as amended following additional or alternative airworthiness requirements.</td>
<td>Master Data List Document MDL-51394-001, Issue 40/Aircraft Flight Manual Supplement FA5-1394-001, Issue 40 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<tr>
<td>10074488</td>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td></td>
<td></td>
<td>737-400</td>
<td>Reduction of Minimum Take Off Weight. Reduced Maximum Take-off Weight for B737-400F to 61,234 Kg.</td>
<td>The Certification Basis for the original product remains applicable without certificate/approval.</td>
<td>Master Data List Document MDL-1394-001, Issue 40/Aircraft Flight Manual Supplement FA5-1394-001, Issue 40 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<tr>
<td>10074496</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
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<td></td>
<td>A340-643</td>
<td>Installation of the FillFlySky Wi-Fi System on board Airbus A340/Airplanes</td>
<td>The Certification Basis for the original product remains applicable without certificate/approval.</td>
<td>340126-RPT02 Master Data List Document Issue 340126-MHB01 Flight Manual Supplement Issue 340126-MHB02 Maintenance Manual Supplement Issue 1340126-MHB02 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<td>10074500</td>
<td>SKIWAY GROUP Inc.</td>
<td></td>
<td>FAA STC SA4178NY</td>
<td>501</td>
<td>Installation of Cabin Entry Step Assembly on Textron Aviation, Inc. Models 501 and 551 aircraft</td>
<td>The Certification Basis for the original product remains applicable without certificate/approval.</td>
<td>1. Aerospace Design &amp; Compliance, LLC. Master Data List Issue C551- MDL-0041 Rev IR(2), dated 16 April 2019 or later FAA approved revisions. 2. Instructions for Continued Airworthiness (CA) C551-ICA-0041, Rev IR(u) dated 10 April 2019 or later FAA accepted revision.</td>
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<td>10074501</td>
<td>WAYPOINT AEROTEC LIMITED</td>
<td></td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>A320-214</td>
<td>Vueling A320 WiFi Connectivity System De-Modification of Aircraft with EASA STC No. 10052556 installed.</td>
<td>The Certification Basis for the original product remains applicable without certificate/approval.</td>
<td>EASA Approved/Waypoint Aerospace Limited Instructions for Continued Airworthiness No.WAL-ICA-104-01 Issue 1/ EASA Accepted/Waypoint Aerospace Limited Master Data List No. WAL-MDL-104-01 Issue 20 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
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<tr>
<td>10074503</td>
<td>T.P.S. S.P.A.</td>
<td>REV. 1</td>
<td>STC Holder</td>
<td>A9139 AB139, A9139 AB139</td>
<td>Installation of midrow interior EMS Kit P/N TPS139-0300-0000-00 This EMS Kit installs inside the main cabin of the Leonardo helicopter AW139 a rack area P/N TPS139-0300-100-00-0 together with one stretcher P/N TPS139-0300-3000-00-0 Rev. 1 Change of the approval holder from TPS AEROSPACE ENGINEER...</td>
<td>The Type and OIS Certification Bases (CB) for the original product scheme applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval...</td>
<td>- Flight Manual Supplement TP1S139-0300-RFMS-00 Master Document List TP1S139-0300-M1100 Master Drawing List TP1S139-0300-M1w11 Instructions for Continued Airworthiness TP1S139-0300-ICA-00 Installation Service Bulletin TP1S139-0300-SBT-0000 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<tr>
<td>10074520</td>
<td>AIRBUS S.A.S.</td>
<td>A320-214</td>
<td>STC Holder</td>
<td>STANDALONE VIUHF SYSTEM ACTIVATION/ MOD AO165: activation of ViUHF System on ACJ320cc.</td>
<td>Demands STC 10074021 The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact on the noise database.</td>
<td>- MAS AO165 Issue 1 AIS-AJA-2016 Issue 02 Certification Summary later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse interaction between this change/repair and any other previously installed change and/or repair. Prior to installation no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10074523</td>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>A321-211</td>
<td>STC Holder</td>
<td>CABIN RECONFIGURATION OF THE PASSENGER CABIN prior to installation of this change/repair it must be determined that there is no adverse interaction between this change/repair and any other previously installed change and/or repair.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact...</td>
<td>- Master Document List S.OAEZ.011639-MDL, Rev. 01 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Airbus S.A.S. MOD No. 151369 &quot;Equipment / Furnishing Pax Compartments of Airbus A321-211: &quot;Men Church&quot; installation on ACJ320cc.</td>
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<tr>
<td>10074577</td>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td>737-800 737-8</td>
<td>STC Holder</td>
<td>INSTALLATION OF TWO HAWKER 1400D SEATED LEAD ACID (SLA) BATTERIES</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact...</td>
<td>Master Document List MEDAVIA D.O. Master Data List MODE0208H (MDL/MEDAVIA D.O.) Instructions for Continued Airworthiness MODE0208H later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>The passenger cabin seats shall ensure a minimum distance of 36 inches between the aft face of the front row monument and the seat reference point of the front row seats. The passenger cabin seating configuration is not part of this design change. Prior to installation of this change/repair...</td>
</tr>
<tr>
<td>10074588</td>
<td>GCA UK RESPONSIBILITY</td>
<td>ATR 72-212 A</td>
<td>STC Holder</td>
<td>INSTALLATION OF TWO HAWKER 240D SEATED LEAD ACID (SLA) BATTERIES</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact...</td>
<td>- MASTER DOCUMENT LIST Issue 1.18. 14th October 2020.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse interaction between this change/repair and any other previously installed change and/or repair. Prior to installation no adverse effect upon the airworthiness of the product.</td>
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### List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<th>Rev. level</th>
<th>Foreign/NAA STC ref</th>
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<th>Certification basis</th>
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<th>System Record</th>
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<tr>
<td>10074589</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>747-8</td>
<td></td>
<td>GSM System Activation</td>
<td>This change describes the activation of a &quot;Global System for Mobile Communication (GSM)&quot; Function of a pre-installed GSMConNX system.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact...</td>
<td>The change is defined with Revision A of AMAC Aerospace Switzerland/Master Data List document MDL-87471-1-3300. The aircraft shall be operated in accordance with Revision A of AMAC Aerospace Switzerland Airplane Flight Manual Supplement documents MDL-87472-1-5300. Or later revisions of the above listed d...</td>
<td>16/10/2020</td>
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<td>10074589</td>
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<tr>
<td>10074596</td>
<td>AV8 PMA, LLC</td>
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<td></td>
<td>HS.125 SERIES 700B</td>
<td>The change is defined with Revision A of AMAC Aerospace Switzerland/Master Data List document MDL-87471-1-3300. The aircraft shall be operated in accordance with Revision A of AMAC Aerospace Switzerland/Airplane Flight Manual Supplement documents MDL-87472-1-5300. Or later revisions of the above listed d...</td>
<td>The equipment components required must be installed prior to the incorporation of this change. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>In accordance with FAA-EASA TIP rev 6, Articles 7.1-1.5, the following statement should be written in the remarks block of the FAA Form 8130-3: &quot;Produced by the holder of the EASA STC number 10074596. Prior to installation of this change/repair it must be determined that the interrelationship between this...</td>
<td>19/10/2020</td>
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<tr>
<td>10074597</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td></td>
<td>767-8</td>
<td>The change is defined with Revision A of AMAC Aerospace Switzerland/Master Data List document MDL-87471-1-3300. The aircraft shall be operated in accordance with Revision A of AMAC Aerospace Switzerland/Airplane Flight Manual Supplement documents MDL-87472-1-5300. Or later revisions of the above listed d...</td>
<td>The equipment components required must be installed prior to the incorporation of this change. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>In accordance with FAA-EASA TIP rev 6, Articles 7.1-1.5, the following statement should be written in the remarks block of the FAA Form 8130-3: &quot;Produced by the holder of the EASA STC number 10074596. Prior to installation of this change/repair it must be determined that the interrelationship between this...</td>
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<td>10074602</td>
<td>SKYWAY GROUP Inc.</td>
<td>FAA STC ST04019NY</td>
<td>500/550/555/560</td>
<td>Installation of Cabin Entry Step Assembly on Textron Aviation, Inc. Models 500, 550, 555 and 560</td>
<td>The change is defined with Revision A of AMAC Aerospace Switzerland/Master Data List document MDL-87471-1-3300. The aircraft shall be operated in accordance with Revision A of AMAC Aerospace Switzerland/Airplane Flight Manual Supplement documents MDL-87472-1-5300. Or later revisions of the above listed d...</td>
<td>The equipment components required must be installed prior to the incorporation of this change. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>In accordance with FAA-EASA TIP rev 6, Articles 7.1-1.5, the following statement should be written in the remarks block of the FAA Form 8130-3: &quot;Produced by the holder of the EASA STC number 10074596. Prior to installation of this change/repair it must be determined that the interrelationship between this...</td>
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<td>10074604</td>
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<td>FMS TO PBN SPECIFICATION (RNP 1)</td>
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<tr>
<td>10074619</td>
<td>AERODATA AG</td>
<td>Installation of Aerodata Flight Inspection System 4D-AFIS-1013</td>
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<tr>
<td>10074623</td>
<td>SKYWAY GROUP Inc.</td>
<td>FAA STC</td>
<td>Installation of Cabin Entry Step Assembly on Textron Aviation, Inc. Models 206, 206A, 206B, and 206H aircraft</td>
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<tr>
<td>10074630</td>
<td>MOTORFLUG BADEN-Baden GmbH</td>
<td>FAA STC</td>
<td>Installation of Direction Finder Antenna/h. Rev. 1: Extension to model BK117 D-2</td>
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<tr>
<td>10074631</td>
<td>AIRWORTHINESS RESOURCES CORPORATION</td>
<td>FAA STC</td>
<td>Installation of a Split Forward Cargo Door Window in accordance with Transport Canada Civil Aviation (TCCA) STC SA20-34</td>
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**EASA STC no.:** 10074604  
**STC Holder:** STC Holder  
**Certification basis:** AIRWORTHINESS Certification basis  
**Description:** FMS TO PBN SPECIFICATION (RNP 1)  

**EASA STC no.:** 10074619  
**STC Holder:** AERODATA AG  
**Model:** B200  
**Description:** Installation of Aerodata Flight Inspection System 4D-AFIS-1013 in Beechcraft King Air B200.  

**EASA STC no.:** 10074623  
**STC Holder:** SKYWAY GROUP Inc.  
**Model:** FAA STC  
**Description:** Installation of Cabin Entry Step Assembly on Textron Aviation, Inc. Models 206, 206A, 206B, and 206H aircraft  

**EASA STC no.:** 10074630  
**STC Holder:** MOTORFLUG BADEN-Baden GmbH  
**Model:** FAA STC  
**Description:** Installation of Direction Finder Antenna/h. Rev. 1: Extension to model BK117 D-2  

**EASA STC no.:** 10074631  
**STC Holder:** AIRWORTHINESS RESOURCES CORPORATION  
**Model:** FAA STC  
**Description:** Installation of a Split Forward Cargo Door Window in accordance with Transport Canada Civil Aviation (TCCA) STC SA20-34.
<table>
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<tr>
<th>STC Holder</th>
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<tr>
<td>CASCADE AEROSPACE Inc.</td>
<td>TCCA</td>
<td>Speed Brake Warning Modification The modification involves the installation of a suplementary speedbrake caution system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Modifiable: yes.</td>
<td>10074633</td>
<td>TCCA SA03-75 ISSUE 4</td>
<td>22/10/2020</td>
<td>Active</td>
<td>10074632</td>
</tr>
<tr>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC SA01050DE</td>
<td>Installation of aircraft main battery replacement with Concorde valve-regulated Lead-Acid battery</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA01050DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Modifiable: yes.</td>
<td>10074673</td>
<td>FAA STC SA-46-600TP</td>
<td>23/10/2020</td>
<td>Active</td>
<td>10074637</td>
</tr>
<tr>
<td>KELOWNA FLIGHTCRAFT Ltd</td>
<td>TCCA</td>
<td>EFB Installation Previsions This modification provides provisions for the installation of interfacing with Electronic Flight Bags (EFB) on B737-300/400/500 aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Modifiable: yes.</td>
<td>10074689</td>
<td>TCCA SA17-17 ISSUE 1</td>
<td>23/10/2020</td>
<td>Active</td>
<td>10074638</td>
</tr>
<tr>
<td>SABENA TECHNICS BGC</td>
<td>A321-</td>
<td>Deactivation of Broadband Satcom System &amp; ARINC 791 Antenna Removal. This modification consists of altering the information described in EASA STC SA01050DE.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Modifiable: yes.</td>
<td>10074643</td>
<td>A321-2711N-2711X-272IN-272SN-</td>
<td>23/10/2020</td>
<td>Active</td>
<td>10074643</td>
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<tr>
<td>AO AVIATION LLC</td>
<td>FAA STC SA01871W</td>
<td>Installation of a Single Point Pressure Refueling System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA01871W. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Modifiable: yes.</td>
<td>10074655</td>
<td>FAA STC SA01871W</td>
<td>30/10/2020</td>
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**List of EASA Supplemental Type Certificates**

**Column Definitions:**
- **STC Holder:** Name of the manufacturer/holder of the STC.
- **Model:** Model number and description.
- **Description:** Details of the modification or change.
- **Certification basis:** Basis for the certification, including applicability and approval requirements.
- **EASA STC no.:** EASA Supplemental Type Certificate number.
- **Date of Issue:** Date when the STC was issued.
- **Status:** Status of the STC (Active, Revoked).

**Links:**
- [EASA STC home page](easa.europa.eu)
- [List of EASA Supplemental Type Certificates](easa.europa.eu)

**Selected Modifications:**
- **Deactivation of Broadband Satcom:** This modification deactivates the broadband satellite communication system.
- **Speed Brake Warning:** The modification adds a speed brake warning system to enhance safety.
- **EFB Installation:** This modification provides provisions for interfacing with Electronic Flight Bags (EFB) on specified aircraft models.

**Other Relevant Information:**
- [Regulations and Standards](easa.europa.eu)
- [Technical Implementation Procedures](easa.europa.eu)
- [Instructions for Continued Airworthiness](easa.europa.eu)
- [Content of STCs](easa.europa.eu)

**Note:** The information provided is a snapshot and may not be exhaustive. For the most up-to-date and comprehensive information, refer to the official EASA STC database.
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<tr>
<td>LUMINATOR HOLDING L.P. dba Luminator Aerospace</td>
<td>FAA STC SR4481AT</td>
<td>SEE FAA ANL</td>
<td>Installation of Luminator Aerospace LS400 Searchlight</td>
<td>The Certification Basis for the original product remains applicable to those certificates/approval, except where amended by additional or later amendments if indicated on FAA STC SR4481AT. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/approval without any impact.</td>
<td>Master Changes UM (MDL) No. MDL-ST-2019-048-02&lt; (&amp;&amp;) Revision B dated 15 June 2020&lt; (&amp;&amp;) or latest FAA approved revision</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. The STC is valid for aircraft types and models approved...</td>
<td>05/11/2020</td>
<td>Active</td>
<td>10071607</td>
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<tr>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A330-281/320-330 A330-281/320-330 A330-381/330-330</td>
<td>INTRODUCTION OF AIRBORNE 4G AIT FEOMTO SITE PROVISIONING</td>
<td>The Certification Basis for the original product remains applicable to those certificates/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the products are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>STC Twenty One Ltd Master Document List No. S21. MDL-1542 Issue 10r later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/10/2020</td>
<td>Active</td>
<td>10074604</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>MBB-BK117 D-2</td>
<td>Night Vision Imaging System (NVIS) installation</td>
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<td>30/10/2020</td>
<td>Active</td>
<td>10074607</td>
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<tr>
<td>AVIACDECO GmbH</td>
<td>737-700</td>
<td>ADS-B Out ModificationThis XPDR modification makes the aircraft fully compatible with EuropeanSPR Rule (EU Regulation 1207/2011 including all amendments published sofar) concerning ADS-Bout. For this purpose the two previously installed Rockwell Collins GPXPDRs have been changed to their successors TP...</td>
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<td>02/11/2020</td>
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<td>10074725</td>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>A330-342/343; A330-321/320/323; A330-321-200/200/223-224</td>
<td>TAIL MUTE WIRE REMOVAL ON GX SYSTEMSwith this STC, the GX Self-Heal Wide Band Broad Band Satellite System (on/offעתה/יתל&quot;מ) Tail Mute hardware protection is removed, and the Tail Mute function is now managed via software functionality only. The Tail Sector Mute function is mainly designed to protect other...</td>
<td>The Certification Basis (CB) for the original product remains applicable/this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>- Master Drawing List ref: 01558-EAS-B-MDL, rev D</td>
<td>- Aircraft Flight Manual Supplement 01558-EAS-B- AFM-01 rev B dated 28OCT-2020. The aircraft Flight Manual Supplement 01558-EAS-B- AFM-01 rev B dated 28OCT-2020. The aircraft Flight Manual Supplement 01558-EAS-B- AFM-01 rev B dated 28OCT-2020. Applicable to aircraft modified per EAD Aerospace STC 10065509 1.KTIV/MH SYSTEM INSTALLATION ON AIRBUS LONG RANGE AIRCRAFT and EAD Modification 01555-EAS-CP/0 to installation of this change/repair it must be determined that the interaction between this change/repair and any other/previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>03/11/2020</td>
<td>Active</td>
<td>10074725</td>
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<tr>
<td>10074761</td>
<td>LUFTHANSA TECHNICH AG</td>
<td>REV. 3</td>
<td>A320-233</td>
<td>RTAF 2: VIP cabin complianceRev. 1: typo correction in model validityRev. 2: CCD and ICA limitations removedRev. 3: ELUpgrade</td>
<td>The Certification Basis for the original product as amended the following additional or/and alternative airworthiness requirements: Special Condition (s): D-2341-001 Pilots and Signage-9-71. The following paragraph(s) at a later amendment: CS 25 App.S25.705(a) at amd 19; CS 25 App.S25.40(a) at amd...</td>
<td>- Aircraft Flight Manual A-00-1989-AMF Issue 01 rev 00- Flight Crew Data A-00-1989-FCD Issue 01 rev 00- Cabin Crew Data A-00-1989-CCD rev 05- Instruction for Continued Airworthiness A-00-1989- ICA Issue 02 rev 02-Master Minimum Equipment List A-00-1989-MMEL rev 00- Supplemental Type Design Doc...</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other/previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>24/05/2022</td>
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<tr>
<td>10074764</td>
<td>SST FLUGTECHNIK GmbH</td>
<td>KODIAK 100</td>
<td></td>
<td>Battery InstallationThis installation is considered as an EASA approved design change/repair to the installation of Daher (former Quest) AircraftDesign LLC. Fast Service Instruction FS-148, Standby Battery System...</td>
<td>The Certification Basis (CB) for the original product remains applicable/this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>- Installation according to the Project Document List document-POL- KOD100-2430-01 Rev. – dated 16/10/2020- Operations according to the Aircraft Flight Manual SupplementAFMS- KOD100-2430-01 Issue 0-0, dated 09/10/2020- Maintenance according to the Aircraft Maintenance ManualSupplementMMILS- KOD100-2430...</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other/previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>05/11/2020</td>
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<tr>
<td>10074771</td>
<td>CESSNA DUESSELDOF CATIONSERVIC E CENTER GmbH</td>
<td>525B</td>
<td>525</td>
<td>Provisions for wireless connection of a PED/EFB to Garmin G3X00 avionics Garmin FS-510...</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements/the following paragraph(s) at a later amendment: Only for model 525 (SN 525-0805; 0800 and on): CS-23 Initial Issue 23.1; 771; 773; 1301; 1309; 1321; 1431; 1501; 1519; 1529...</td>
<td>- Cessna Düsseldorf Citation Service Center GmbH, Master Document List,Document No. 2020-24-16, Rev IR, dated 22 October 2020. To be installed in accordance with Installation Procedure, Document No 2020-24-1, Rev IR, dated 18 September 2020. To be operated in accordance with AFM-5, Document No. 2020-24...</td>
<td>Limited to the following serial numbers:525: 525-0805, 0800-0805(52); 525B-0057; 0451-...on/offset: 525B-0057; 0451-...on/off/previously installed change and/or repair will introduce no adverse affect upon th...</td>
<td>06/11/2020</td>
<td>Active</td>
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<tr>
<td>10074772</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>747-8</td>
<td>Installation of C-Music System self defense system. The change includes the Pod Structural Assembly. (PSA) housing the maincomponents of the system. The PSA is fuselage-mounted at the lowersection of the fuselage.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s): B-101, Vibration / buffeting/ compliance criteria for large external antenna installations/The requirements for environmental protection and the associated...</td>
<td>The change is defined with Revision D of Master Data List Document/MDL- B7473-1-500. The aircraft shall be operated in accordance with Revision A of Aircraft Flight Manual Supplement AFMS- B7473-1-500. The aircraft shall be maintained in accordance with Revision B of Instructions for Continued Airworthi...</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other/previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>06/11/2020</td>
<td>Active</td>
<td>10074772</td>
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<td>STC Holder</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>GROWING ENGINEERING PARTNERS GmbH</td>
<td>Rev 1</td>
<td>A330-200</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact...</td>
<td>Master Document No. S.OAEZ.001935-MDL, Rev. 0.</td>
<td>Instruction for Continued Airworthiness S.OAEZ.001935-ICA, Rev. 0/0/0 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Cable Layout STC 100485S is a pre-requisite for installation. Lufthansa Technik EASA STC 1004507 Rev.0/1. &quot;Installation on/off-electrical Floor Path Marking System&quot; is a pre-requisite for installation. Prior to installation of this change/repair it must be determined that the interrelationship between the...</td>
<td>21/04/2021</td>
<td>Active</td>
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<tr>
<td>SWS CERTIFICATION SERVICES Ltd.</td>
<td></td>
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<td></td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact...</td>
<td>Master CONFIGURATION SCHEME. doc. Ref. 3213-MCS-01, issue 3 dated 02/03/2020(+v)...) later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Applicable to A350-941 modified per SWS EASA STC 1005697/1. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/11/2020</td>
<td>Active</td>
<td>10074808</td>
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<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td></td>
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<td></td>
<td>Installation of a Constant Friction Mount, Cradle, and Integrated PowerSupply on Boeing 737 Series Aircraft.</td>
<td>1. Avionics Support Group, Inc. Master Data List 213136-MDL, Rev. W dated 12 April 2018, or later FAA approved revision. 2. Instructions for Continued Airworthiness document 213136-ICA, Rev. C dated 29 November 2017, or later approved revision 3. Airplane Flight Manual Supplement 213131-RMS dated 18...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2020</td>
<td>Active</td>
<td>10074814</td>
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<td>NORTHWEST AEROSPACE TECHNOLOGIES, INC.</td>
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<td></td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS25 art 0: 25.853. The requirements for environmental protection and the associated certification noise and/or emissions level...</td>
<td>Northwest Aerospace Technologies, Inc. (NAT) Master Drawing List (MDL) N0560-D10, Revision D dated 24 January 2017, or later Federal Aviation Administration (FAA) approved revision instructions for Continued Airworthiness (ICA) N0560-D70, Revision A dated 15 January 2017, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is applicable to A350-941 modified per FAA STC STA4024CH installed...</td>
<td>21/01/2021</td>
<td>Active</td>
<td>10074835</td>
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<td>STC Holder</td>
<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AMEC AEROSPACE SWITZERLAND AG</td>
<td>767-400ER</td>
<td>2ku-Band Antenna installation. This change describes the installation of a 2ku Band (GMSK) antenna on the top of the fuselage.</td>
<td>The Certification Basis for the original product as amendability the following additional or alternative airworthiness requirements: Equivalent Safety Finishing(s): B-101 Vibration / buffeting compliance criteria for large external antenna installation. The requirements for environmental protection and the.</td>
<td>The change is defined within Revision C of Master Data List Document MIL 87671-1-5300. The aircraft shall be operated in accordance with Revision A of Aircraft Flight Manual Supplement document AFMS-87671-1-5300. The aircraft shall be maintained in accordance with Revision A of instructions for Continued Airworthiness CE141004019, Revision 1, dated August 3, 2016. * Airplane Flight Manual.</td>
<td>17/11/2020</td>
<td>Active</td>
<td>10074830</td>
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</table>
| GULFSTREAM AEROSPACE CORPORATION | GV, GIV | Installation of a SATCOM Direct Router on Gulfstream Models: G-IV and GIV | Per FAA STC ST04032AT-DThe requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. The requirements applicable to this certificate approval without any impact on the noise database. | Supplemental Type Certification (STC) ST040032AT-D, Installation SATCOM Direct Router (SDR), Date Amended August 25, 2016 Index List ref. CFS526502 Revision B, dated November 5, 2019. Model G-IV. 
<p>| CAA UK RESPONSIBILITY | 777-300ER, 777F, 777-200, 777-200R, 777-300 | BOEING 777 SKYCOUCH SEAT DEACTIVATIONThe STC deactivates 10-off Satellite Seats US (SSU) 5751 Series Sky Couches previously installed under SWS EASA STC 10050705. To accomplish this, SkyCouch legrests &amp; occupant restraint (Cuddle Straps) are being removed and new armrests and theatre seat containers (in... | The Certification Basis for the original product as amendability the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment 25.705(c)(3)(i)shall be amended: The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. The requirements applicable to this certificate approval without any impact on the noise database. | Master Configuration Scheme ref. 3214-MCS-01 Issue 01 dated14/06/2024 (&gt;)or later revisions of the above listed document(s) approved acceptand equipe the EASA system. | 18/11/2020 | Active | 10074860 |
| JAMCO AMERICA, INC. | FAA STA14855E | Upper deck stowage bin and closet modification in accordance with JamcoAmerica Inc. Master Drawing List IA-104 MDL, revision B dated 08/08/94 later FAA approved revision. | - | | 29/11/2004 | Active | 10074883 |
| LUFTHANSA TECHNIK AG | LBA STC TA0352 | VIP cabin completion/Revision: Replacement of conductive oxygen flex-hoses without oxygen leaks. Original LBA certificate number TA0352 modified. | The Certification Basis for the original product as amendability the following additional or alternative airworthiness requirements: Equivalent Safety Finishing(s): D-03 ‘‘Side facing dividers’’ issue 10Revision(s): D-01 ‘‘Direct view’’ issue 1, D-02 ‘‘Doors between passenger compartments’’ issue 1, D-04 ‘‘Firm ha... | As per AFM supplements A-25-00-303, A-25-00-304, A-25-00-305 rev 1 and/or A-00-00-305PFR rev installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product. | 19/11/2020 | Active | 10074896 |
| AEROTEC &amp; CONCEPT | EC 155 B, EC 155 B1 | New CVFDR Installation in accordance with Aerospatiale Change N° 19138707s STC consists in the installation of a combined voice &amp; data flight recorder CURTIS CURTIS MPRF, Flight DataAcquisition Unit (FDAU). | The Certification Basis for the original product as amendability the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: CS 29 Amdt 7, 29, 1316, 29, 1317, 29, 1467, 29, 1469/FAIR 29 Amdt 16: 29, 561. The requirements for environmental protection and o... | Certification Programme AE19136P/00 Rev 1Master Data List AE19136ML Rev 1Instruction for Continued Airworthiness AE19136CA Rev OfforbitFlight Manual AE19136RBF Rev 1 later revisions of the above listed document(s) approved acceptand equipe the EASA system. | 19/11/2020 | Active | 10074897 |</p>
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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10074910</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td></td>
<td></td>
<td>A330-203</td>
<td>Connecting the inflight entertainment system (Zodiac inflight innovationRAVE) to air traffic service / communication management unit in order to provide connecting gate messages to passengers and free text capabilities to cabin crew via aircraft communications addressing and reporting system (ACARS).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Societe Air France S.A. Change Approval Sheet CAS-A330-19-005</td>
<td>In case of Air Traffic Service Unit Part Number evolution (hardware/or software), the operator shall inform Societe Air France S.A. to check the need for a new or amended cybersecurity analysis.</td>
<td>20/11/2020</td>
<td>Active</td>
<td>10074910</td>
</tr>
<tr>
<td>10074911</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.</td>
<td>REV. 1</td>
<td></td>
<td>A321-272NX</td>
<td>Airbus A320 Series - ADS-B OUT v2 InstallationThis STC replaces the existing ATC transponders with new RockwellCollins TPR-901 transponders that are compliant with ADS-B Out DO-260B functions and configures them for ADS-B Out operation Rev.1: Transfer of STC</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>ATL AFM Supplement No. 13651-AFM-00001 Issue 1ATL Instructions for Continued Airworthiness No. 13651-ICA-00001 Issue 1ATL MMEL No. MMELS-13651-00001 Issue 1ATL Master Document List No. MDL-13651-00001 Issue 1 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>STC No.</th>
<th>STC Holders</th>
<th>Rev. level</th>
<th>Models</th>
<th>Description</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>10074937</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td>A330-343, A330-342, A330-343, A330-323, A330-322, A330-321, A330-300, A330-302, A330-301, 300-310, 310-321, 322</td>
<td>Wi-Fi/Wi4ES Installation and Installation of Wi-Fi system into the cabin of the A330 family for passenger usage</td>
<td>The certification basis for the original product remains applicable to the certificate/approval. The requirements for environmental protection and the associated noise/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td></td>
<td></td>
<td>AFMS NAVO021 R10 (or later revisions of the above listed document(s) approved/accepted) and the EASA system.</td>
<td>24/11/2020</td>
<td>Active</td>
<td>10074937</td>
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<tr>
<td>10074939</td>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td>FAA STC ST04614T</td>
<td>737-900ER, 737-900, 737-800, 737-700, 737-600, 737-500, 737-400, 737-300</td>
<td>Installation of the Honeywell Solid State Cockpit Voice Recorder (SSCVR) with ACARS message recording capability</td>
<td>The certification basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraphs at a later amendment CS 25.1250 CS 25.1301 and CS 25.1529: The requirements for environmental protection and the associated noise/ or emissions levels of the on...</td>
<td></td>
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<td>1. Master Drawing List, dated 28 March 2019, or later FAA approved revision 2, Master Drawing List, dated 28 October 2020, or later FAA approved revision 2, Master Drawing List, dated 1 August 2019 or later FAA accepted revision.</td>
<td>17/12/2020</td>
<td>Active</td>
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<tr>
<td>10074945</td>
<td>AMS HELI DESIGN, LLC</td>
<td>FAA STC SR09917AC</td>
<td>AW109SP, A109S</td>
<td>Installation of an Advanced Life Support System (ALS) airborne EMS ambulances interior/Revision 1 of the FAA STC reference.</td>
<td>The certification basis for the original product remains applicable to the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SR09917AC: The requirements for environmental protection and the associated noise/ or emissions levels of the on...</td>
<td></td>
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<td>AMS Master Drawing List Document No. 1835950301, Revision F, dated 26 October 2020, or later FAA approved revision 2, Master Drawing List Document No. 183592001, Revision A, dated 29 September 2020, or later FAA approved revision, is required for the A109S and Rotorcraft Flight Manual Sup...</td>
<td>10/08/2021</td>
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<tr>
<td>10074953</td>
<td>BELL HELICOPTER, TEXTRON, Inc.</td>
<td>FAA STC SR11244DS</td>
<td>S05</td>
<td>Installation of Expanded Instrument Panel (EIP)</td>
<td>The certification basis for the original product remains applicable to the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SR11244DS: The requirements for environmental protection and the associated noise/ or emissions levels of the on...</td>
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<td>Master Drawing List, Report No. AA-17015, Revision A, dated 7th November 2019, or later FAA approved revision, is required for the A109S and Rotorcraft Flight Manual Sup...</td>
<td>24/11/2020</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10074957</td>
<td>เทเลดีน ฮัมเบิร์ก</td>
<td>B300C B300 B200GT B2000GT 200C 200</td>
<td>Installation of GPS Antenna installation of one GPS Antenna with antenna doubler and coaxial cable.</td>
<td>The original certification basis is complemented by CS 23 all initialises as documented in certification plan 617-10017-CPL-0001 Rev. C. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this c...</td>
<td>Installation in accordance with Master Document List, 617-10017-MDL-0001 Rev. A, with detailed instructions in document 617-10017-CA001 Rev. B or later revisions of the above listed document(s) approved...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/11/2020</td>
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<td>10074957</td>
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<td>10074959</td>
<td>Boeing Ireland Limited</td>
<td>737-800</td>
<td>SatLIFK antenna removal.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any ...</td>
<td>Change/Repair document BL-00175-A009 Rev 00 for installation of additional antennas. BL-00175-CA01 Rev 00 Airworthiness Limitations Manual Supplement BL-00175-M003 Rev or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>- the installation of an antenna as per STC 10064143 is a prerequisite to the minor change BIL-00174 for deactivation and removal of ArcheionSatLIFK system is a prerequisite...</td>
<td>25/11/2020</td>
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<tr>
<td>10074962</td>
<td>SAA Solutions for Aviation, S.L.</td>
<td>C-212-VA C-212-DE, C-212-DF, C-212-EE C-212-CF, C-212-DD C-212-CB, C-212-CC, C-212-CD</td>
<td>DS-B Out and PBN Avionics Systems Upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any ...</td>
<td>A1219-RPT10 Master Document List Edition 3A/212/HK01-A Flight manual supplement Edition 2A/212/HK01-B Flight manual supplement Edition 2A/212/HK01-C Flight manual supplement Edition 2A/212/HK02 Maintenance Manual Supplement Edition 20 Or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Sa A1219-HD01-A Flight manual supplement Ed2, A1219-HK01-B Flight manual supplement Ed2, A1219-HK01-C Flight manual supplement Ed2 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/11/2020</td>
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<td>10074963</td>
<td>Luftansa Technik AG</td>
<td>LBA TA0745</td>
<td>747-400</td>
<td>Installation of a VIP cabin interior “El Meymon”, modified with an aisleway from passenger deck to lower lobe cargo compartment. Revision: replacement of conductive oxygen flex-hoses with non-conductive ones.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements CRF A-1 “Certification Basis” Issue 2. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any ...</td>
<td>Generic documents Supplemental type design document STDG 25-0219M issue 01 Rev 00 Original installation documents ATM supplement A-00-0317-AFM rev IPAAM supplement A-00-0317-AFM rev IRAAM supplement A-00-0317-AFM Rev 00 for installation of the above listed document(s) approved/accepted under the EASA system.</td>
<td>As per AFM supplement A-00-0317-AFM Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/11/2020</td>
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<td>10074980</td>
<td>Testedyne UK Limited</td>
<td>REV. 1 CAA UK RESPONSIBILITY</td>
<td>757-200</td>
<td>Installation of Teledyne Controls Wireless Ground Quick Access Recorder (WQAR) P/N 2243800. Rev. 1. Transfer from Teledyne Limited to Teledyne UK Limited.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any ...</td>
<td>Teledyne Limited Modification FDC M011 (Master Data List/Definition Document STC-MDL-011-03) dated 15/12/2004 (or) later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/11/2020</td>
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<td>LEARJET Inc.</td>
<td>Installation of a Vased Ka Radar</td>
<td>10075065</td>
<td>Transfer of STC 10075065</td>
<td>Standard Aerotechnik Master Data List 10038141, Revision (E), dated 13 November 2020Instructions for Continued Airworthiness, Document 1038100, Revision(A) dated 13 November 2020or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implement...</td>
<td>30/06/2021</td>
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<td>SPECIALIST AVIATION SERVICES LIMITED</td>
<td>T-PED Approval in accordance with FAA STC 10075005</td>
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<td>Flight Manual Supplement FMS/1018-043 Issue AT-PED Compatibility Report (ICA) DR/1018-043 Issue SIService Bulletin SI 23-119 Issue BDeclaration of Compliance of Issue B of SAMS/M0110B or later revisions of the above listed document(s) approved/accepted on behalf of EASA system.</td>
<td>04/12/2020</td>
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<td>10770304</td>
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<td>IKHANA AIRCRAFT CANADA, Inc.</td>
<td>Enhanced Air Conditioning System Installation in accordance with Transport Canada Civil Aviation (TCCA) STC SA16-33.</td>
<td>1007544</td>
<td>Ikhana Aircraft Services Manufacturing Drawing List ERS210-2016, Revision(A), dated February 26, 2019$Ikhana Aircraft Services Installation Drawing List ER219-2017, Revision(A), dated February 19, 2019$Ikhana Aircraft Services Installation Drawing List ER218-2017, Revision(A), dated February 18, 2019$Ikhana Aircraft Services Installation Drawing List ER217-2017, Revision(A), dated February 17, 2019$Ikhana Aircraft Services Installation Drawing List ER216-2017, Revision(A), dated February 16, 2019</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>AEROLITE AG</td>
<td>Installation of EMS Equipment P/N 145060-501.</td>
<td>1007560</td>
<td>- Aerolite Max Busch AG Master Drawing List DWL-301-Rotorcraft Flight Manual Supplement FMS-301-Operation and Maintenance Manual OPM-301 or later revisions of the above listed document(s) approved/accepted on behalf of EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/12/2020</td>
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<td>2 EXCEL DESIGN LIMITED</td>
<td>Introduction of Osprey 50 in LEAP-R.His design change introduces an Osprey 50 Radar system and Interconnect between the Leading Edge Application PodRadar (LEAP-R) approved under design change 2XL020 STC (10062193).</td>
<td>1007568</td>
<td>- Installation of Osprey 50 in LEAP-R in accordance with Leading Edge 2Excel Navigation Master Document List: 2XL049, at issue 2, May 14, 2020-Operation in accordance with Aircraft Flight Manual SupplementLeading Edge 2Excel Navigation Document 2XL049 LEFT 7(11), at Revision 1, dated December 1, 2021, ...</td>
<td>The following Leading Edge, 2Excel Aviation changes constitute approval-requisite for the installation: 2XL019, 2XL020, 2XL021, 2XL030, 2XL051, 2XL059, 2XL044, 2XL040 without role of equipment.Prior to installation of this change/repair it must be determined that the interrelationship between this change...</td>
<td>03/12/2020</td>
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<td>ATELIER CRABORD</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: 23.65, 23.69, 23.71, 23.77, 23.201, 23.65, 23.69, 23.71, 23.77, 23.201, 23.203.</td>
<td>10075077</td>
<td>- Master Drawer List MDL-TB20-16-01 dated 19 November 2020-Aeroplane Flight Manual Supplement / Suplement au Manuel du Vol Ref.SUPPMV TB200-16-01(a), (b), (c), (d), dated 10.12.2020 or later revisions of the above listed document(s) approved/accepted on behalf of EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>DAVIDSON FALCON JET CORPORATION</td>
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<td>FAA ST04217CH</td>
<td>FALCON 2000EX FALCON 2000</td>
<td>EGPR/AESenhanced Ground Proximity Warning System upgrade (238-238) with allsmartRunway SmartLandingoption (RAAS)</td>
<td>- MDL 1032202 Rev C</td>
<td>AFMS 1032005 Rev B - ICA 1032661 Rev D-MMEL Supplement DJ-MMELS - F2000-0011 issue 1: MMEL Supplement DJ-MMELS - F2000-0011 issue 1: for later revisions of the above listed document(s) approved/accepted on behalf of EASA/An acceptance with the Technical Implementation/Procedures of the The 'Reparable Customer Definitions (RCD)’ part number that is based on the Enhanced Ground Proximity Computer shall be: 718-1601-203 (Configuration 1, Low Volume). Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any others...</td>
<td>04/12/2020</td>
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<td>10775075</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>MBB-BK117 D-3 MBB-BK117 D-2</td>
<td>Installation of Flightdeck DZMs with optional activation of integrated Wireless Functions (WIFI / Bluetooth) Revision 1/Extension of applicability to MBB-BK117 D-3 and addition of an external GSM Antenna location and introduction of label/D2M KAVFW</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>Revisions 9RF5MS for Flightdeck DZMs, Doc. 2117706-AM A issue4 dated 24.11.2021/ Addendum to RFMS for Flightdeck DZMs, Doc. 2117706-AM B issue 2 dated 24.09.2020/MMEL Supplement, Doc. 2117706-ME Issue 1 dated 21.01.2020/Instructions for Continued Airworthiness, Doc. 2117706-CA Issue 1 dated 26.05.2020.</td>
<td>MBB-BK117 D-2m is excluded. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other preexisting installed/installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>03/06/2024</td>
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<td>GDC ENGINEERING GmbH</td>
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<td>A320-251N</td>
<td>A320-251N cabin modification</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>Master Certification Document List CEN-MCDO-A320B Issue 01 dated 4 Dec 2020/CA &amp; Operator Manual Supplement A320B-ICA Rev A dated 3 Dec 2020/ revisions of the above listed document(s) approved/accepted/updated the EASA system</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other preexisting installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>10/12/2020</td>
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<td>AEARO TECHNOLOGIES LLC</td>
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<td>FAA STC ST04387CH</td>
<td>FALCON 2000, FALCON 2000EX</td>
<td>Install thermal/acoustic insulation system</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by FAA STC ST04387CH.The requirements for environmental protection and the associated certified noise and/or emissions levels of the original...</td>
<td>Aearo Technologies Master Data List DML-1794-002. Revision A, dated 15 January 2019, or later FAA approved/accepted revisions/FAA/accepted Instructions for Continued Airworthiness, Manual/C-1794-001, Revision IR, dated 17 January 2019, or later FAA acceptance/revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other preexisting installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>08/12/2020</td>
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<td>ROCKWELL COLLINS, Inc.</td>
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<td>FAA ST01502W</td>
<td>747-400, 747-400F</td>
<td>Modification of an existing Rockwell Collins Global Landing Ini (G-LiiMulti Mode Receiver (MMR), Sensor Systems Inc. GPS Antennas and ACSTransponder</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by FAA STC ST01502W.The requirements for environmental protection and the associated certified noise and/or emissions levels of the original...</td>
<td>1. Master Drawing List RCA-6037, Revision (-) dated 15 January 2019/2. Instructions for Continued Airworthiness RCA-6204, Revision (-) dated 12 December 2018, or later FAA approved/accepted revision for item 1 and 2.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other preexisting installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>25/01/2021</td>
<td>Active</td>
<td>10775109</td>
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<td>LEE AEROSPACE, Inc.</td>
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<td>FAA STC ST01238W</td>
<td>SEE FAA AML</td>
<td>Replacement of OEM window pane assemblies with Lee Aerospace laminated/heat resistant glass and termination of OEM dry air system</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by FAA STC ST01238W/The requirements for environmental protection and the associated certified noise and/or emissions levels of the original...</td>
<td>Lee Aerospace Master Drawing List IAI-001-MDL, Revision 0, dated 13 February 2004, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other preexisting installed change and/or repair will introduce no adverse affect upon the airworthiness of the product. The STC is valid for aircraft types and models approved...</td>
<td>08/12/2020</td>
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<td>LEE AEROSPAC, INC.</td>
<td>10075112</td>
<td>FAA STC STO1938W</td>
<td>MU-30-400A/450</td>
<td>Installation of Windshields</td>
<td>The Certification Basis: The original product remains applicable to this certificate/approval, except where amended by additional or later amendments. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged. Any change/repair must be determined with the EASA.</td>
<td>- For model MU-300 Master Data List, No. 17-003-SC4351 Rev1, dated 4 December 2019.</td>
<td>Prior to installation of this change/repair it must be determined with the EASA system.</td>
<td>08/12/2020</td>
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<td>10775132</td>
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<td>LEARJET Inc.</td>
<td>10075118</td>
<td>FAA STC STO4466CH</td>
<td>BD-100-1A10</td>
<td>Installation of a Viasat Business Aviation Router (VBR) P/N137138</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged.</td>
<td>Master Data List 1039404, Revision (B), dated 13 November 2020.</td>
<td>Prior to installation of this change/repair it must be determined with the EASA system.</td>
<td>06/12/2022</td>
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<td>LEARJET Inc.</td>
<td>10075120</td>
<td>FAA STC STO4467CH</td>
<td>BD-100-1A10</td>
<td>Installation of a Viasat Ka Terminal Rev. 1.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged. Any change/repair must be determined with the EASA.</td>
<td>- StandardAero Master Data List 1010974, Revision (B), dated 13 November 2020.</td>
<td>Prior to installation of this change/repair it must be determined with the EASA system.</td>
<td>30/06/2021</td>
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<td>SPECIALIST AVIATION SERVICESLMITE D</td>
<td>10075124</td>
<td>FAA STC SA02658SE</td>
<td>SA02658SE</td>
<td>Installation of Garmin GI 275 Multi-Function Electronic FltDisplay</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged. Any change/repair must be determined with the EASA.</td>
<td>- For Aircraft in FAA AML to the FAA AML - Master Data List, No. 1016-02246-15, Revision 4, dated 4 December 2023.</td>
<td>Prior to installation of this change/repair it must be determined with the EASA system.</td>
<td>09/12/2020</td>
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<td>GARMIN INTERNATIONAL, Inc.</td>
<td>10075142</td>
<td>FAA STC SA02558SE</td>
<td>SA02558SE</td>
<td>Installation of Garmin GI 275 Multi-Function Display System/Rev. STC validation for Garmin GI 275 Multiple Function Electronic Display Rev. 2.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged.</td>
<td>For Aircraft in FAA AML to the FAA AML - Master Data List, No. 1016-02246-15, Revision 4, dated 4 December 2023.</td>
<td>Prior to installation of this change/repair it must be determined with the EASA system.</td>
<td>25/03/2024</td>
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<td>JET AVIATION AG</td>
<td>10075151</td>
<td>A320-214</td>
<td>A320-214</td>
<td>Installation of Windshields</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative noise and emissions requirements: See Annex dated 25/03/2024 The EASA System.</td>
<td>For Aircraft in FAA AML to the FAA AML - Master Data List, No. 1016-02246-15, Revision 4, dated 4 December 2023.</td>
<td>Prior to installation of this change/repair it must be determined with the EASA system.</td>
<td>18/01/2021</td>
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<td>BD-700-1A11</td>
<td>BD-700-1A11</td>
<td>Installation of Aviation Clean Air Bi-Polar Ionization System</td>
<td>Duncan Aviation, Inc. STC Master Document List Doc. No. CERTDOC-011232 Rev. A, dated 12 August 2020, or later FAA approved revisions. Duncan Aviation, Inc. Instructions for Continued Airworthiness, Doc. No CERTDOC-001133, Rev. B, dated 12 August 2020, or later FAA accepted revisions.</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/12/2020</td>
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<td>FAA STC 103338AT</td>
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<td>CL-600-2E25</td>
<td>(RU SERIES 100)</td>
<td>Installation of LRU Avionics Protective CoverRev. 1 extends the applicability to model CL-600-2E25.</td>
<td>- Chippewa Aerospace Master Data List (MDL) No. 601-2451-01-0001 Rev. 9, dated 21 October 2019, - Instructions for Continued Airworthiness (CAA), Chippewa Aerospace, Inc. Document No. 601-2451-01-0002 Rev. 6, dated 08 April 2019 or later revisions of the above listed document(s) approved/ accepted...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/02/2021</td>
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<td>FAA STC 109935AT</td>
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<td>CL-600-2D24</td>
<td>(RU SERIES 90C)</td>
<td>Installation of LRU Avionics Protective CoverRev. 1 extends the applicability to model CL-600-2E25.</td>
<td>- Chippewa Aerospace Master Data List (MDL) No. 601-2451-01-0001 Rev. 9, dated 21 October 2019, - Instructions for Continued Airworthiness (CAA), Chippewa Aerospace, Inc. Document No. 601-2451-01-0002 Rev. 6, dated 08 April 2019 or later revisions of the above listed document(s) approved/ accepted...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10 Dec 2020</td>
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<td>FAA STC 104181AT</td>
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<td>CL-600-2D15</td>
<td>(RU SERIES 700)</td>
<td>Installation of LRU Avionics Protective CoverRev. 1 extends the applicability to model CL-600-2E25.</td>
<td>- Chippewa Aerospace Master Data List (MDL) No. 601-2451-01-0001 Rev. 9, dated 21 October 2019, - Instructions for Continued Airworthiness (CAA), Chippewa Aerospace, Inc. Document No. 601-2451-01-0002 Rev. 6, dated 08 April 2019 or later revisions of the above listed document(s) approved/ accepted...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10 Dec 2020</td>
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<td>FAA STC 104181AT</td>
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<td>CL-600-2C10</td>
<td>(RU 700, 701, 702)</td>
<td>Installation of LRU Avionics Protective CoverRev. 1 extends the applicability to model CL-600-2E25.</td>
<td>- Chippewa Aerospace Master Data List (MDL) No. 601-2451-01-0001 Rev. 9, dated 21 October 2019, - Instructions for Continued Airworthiness (CAA), Chippewa Aerospace, Inc. Document No. 601-2451-01-0002 Rev. 6, dated 08 April 2019 or later revisions of the above listed document(s) approved/ accepted...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10 Dec 2020</td>
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### List of EASA Supplemental Type Certificates

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<td>10075198</td>
<td>GAMA AVIATION (ENGINEERING) LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A34</td>
<td>A34</td>
<td>FMS Upgrade with Universal Avionics UNS-1Lw</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>According to Gama Aviation (Engineering) Ltd MDO Document GAEL-1670 MDL-01 Issue 1 to be operated in accordance with Gama Aviation (Engineering) Ltd AFMS Document No. GAEL-1670 AFMS-01 Issue 3 to be installed i.e. Gama Aviation (Engineering) Ltd Document GAEL-1670 DS-01 Issue 1 to be maintained i.e. Gama...</td>
<td>Conditions: This change is only for aircraft equipped as defined in Chapter 3 of GAMA Design Statement GAEL-1670 DS-01. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change...</td>
<td>15/12/2020</td>
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<td>10075201</td>
<td>AIRPLUS MAINTENANCE GmbH</td>
<td>REV. 1</td>
<td>SEE EASA APPROVED MODELS LIST</td>
<td>DR-400/0P DR-400/0NL DR-400/0G DR-400/0G-2 DR-400/200 R DR-400/180 S DR-400/180 R DR-400/180 DR-400/160 D DR-400/160 DR-400/140 B DR-400/140 DR-400/125 DR-400/125 DR-400/120 D DR-400/120 A DR-400/120 DR-400/100 DR-380 DR-360 DR-340 DR-315 DR-300/180 R DR-300/140 DR-300/125 DR-300/120 DR-300/108</td>
<td>Installation of Aspen EFD1000 Primary Flight Display System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) is/are at a later amendment: CS-23, Amendment 41, 29, 301, 303, 305, 307, 308, 309, 1321, 1329, 1335, 1351, 1365, 1431; 152...</td>
<td>- Aircraft Flight Manual Supplement PO-1868-00000, Issue 01- Maintenance Manual Supplement MM-1868-00000, Issue 01 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effects upon the airworthiness of the product.</td>
<td>21/12/2020</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>LBA STC TA0560</td>
<td>737-800</td>
<td>VIP Cabin Completion (EAGLE Rev 1): Replacement of conductive oxygen flex-hoses with non-conductive ones.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): D-1.1 Cabin Conversion (BBU) for Boeing 737-800, Special Condition (s): JAR 25.785(h)(1); JAR 25.813(c)(1); JAR 25.825(c)(1)</td>
<td>General documents: STD-00-02/02 M Issue 03 rev 01 OG installation documents Airplane Flight Manual Supplement Doc. No A-52-03/046-AFM, Rev. IR (July 04/03): Passenger Cabin Internal Doors indication...</td>
<td>- Limitations as per AFM supplements A-52-03/046-AFM, A-26-02/039-AFM, A-52-03/046-AFM and A-52-03/027-AFM: Possible electromagnetic or other interferences between the Head-Up Guidance System and this STC installation have not been checked. Prior to future activation of the Head-Up Guidance System, it is advisable...</td>
<td>16/12/2020</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>LBA STC TA0585</td>
<td>737-800</td>
<td>VIP – Interior and Systems Installation Rev 1: Replacement of conductive oxygen flex-hoses with non-conductive ones. Remunerating certificate from German LBA STC TA0585 Rev 2: typo corrected</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): D-1.1 Cabin Conversion (BBU) for Boeing 737-800, Special Condition (s): JAR 25.785(h)(1); JAR 25.813(c)(1); JAR 25.825(c)(1)</td>
<td>General documents: STD-00-01/10 M Issue 04 rev 02 OG installation documents Lufthansa Technik Specification Doc. No 25-05-017 dated Dec. 31, 20006, 10 revision Supplement to AFM 04/02/024-FDS as follows...</td>
<td>- Limitations as per AFM supplements A-23-01/136-AFM, A-00-02/140-AFM, A-26-02/149-AFM, A-00-02/150-AFM, A-00-02/153-AFM</td>
<td>21/01/2021</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td>LBA STC TA0585</td>
<td>737-800</td>
<td>VIP Cabin Completion (MDRDC Rev 1): Replacement of conductive oxygen flex-hoses with non-conductive ones. Remunerating certificate from EASA A-S 00221.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): D-1.1 Cabin Conversion (BBU) for Boeing 737-800, Special Condition (s): JAR 25.785(h)(1); JAR 25.813(c)(1); JAR 25.825(c)(1)</td>
<td>General documents: STD-00-03/10 M Issue 02 rev 02 OG installation documents Airplane Flight Manual Supplement Doc. No A-00-04/106-AFM, Rev. IR (July/04/03): EDF (Cut Off Switch), dated March 23, 2002, 10 revision...</td>
<td>- As per Airplane Flight Manual Supplement Doc. No A-00-04/106-AFM, RevIRr - Maximum Passenger seating capacity: 25. Prior to installation of this change/repair it must be determined if there is any interference between this change/repair and any other previously installed change and/ or repair will...</td>
<td>16/12/2020</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>CAA RESPONSIBILITY</td>
<td>227-200F</td>
<td>227-200</td>
<td>ACTIVATION OF CPOLC (ATN B1 AND FANS 1A+): AND ESSENTIAL SAT/VOICE FORAT/05This change is to comply with the 2020 mandates for CPOLC (ATN B1) and FANS 1A+. It also facilitates approval of a Long-Range Voice Communications System (LRVC) for Air Traffic Services (ATS) Voice.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): 21-CS25.737 (a)(c)CS25.777(b)(a) CS25.1301 (a)(c) CS25.1304 (a) CS25.1306 (a) CS25.1309 (a) CS25.1309 (b) CS25.1309 (c) CS25.1310</td>
<td>EASA Approved STC Twenty One Ltd Flight Manual Supplement No. S21.FMS-0163 Issue 258781 - As per Airplane Flight Manual Supplement Doc. No A-00-04/106-AFM, RevIRr - Maximum Passenger seating capacity: 25. Prior to installation of this change/repair it must be determined if there is any interference between this change/repair and any other previously installed change and/ or repair will...</td>
<td>17/12/2020</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>LBA STC TA0501</td>
<td>737-800</td>
<td>Installation of Lower Vip Cabin Conversion (Revision 1): Replacement of conductive oxygen flex-hoses with non-conductive ones. Certificate remunerating from German LBA STC TA0501 Revision 1: Limitations in AFM Supplements updated.</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The requirements for the associated cabin noise and/or emissions levels of the product are unchanged and remain applicable to...</td>
<td>General documents: STD-00-01/05/M issue 04 revision 00 OG installation documents Airplane Flight Manual Supplement Doc. No A-23-02/039-AFM Rev IR(Plane Flight Manual Supplement Doc. No A-26-02/225-AFM Rev IR(Plane Flight Manual Supplement Doc. No A-26-02/225-AFM Rev IR(Plane Flight Manual Sup...</td>
<td>Limitations as per AFM supplements A-23-02/226-AFM, A-26-02/230-AFM and A-52-02/226-AFM Prior to installation of this change/repair it must be determined if there is any interference between this change/repair and any other previously installed change and/ or repair will...</td>
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<td>Associated Technical Documentation</td>
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<td>VENTURA AEROSPACE, Inc.</td>
<td>Installation of 8y Rigid Cargo Barrier in accordance with ALPINE TCCA STC 10075254</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval, except where amended by additional or later amendments. The requirements for environmental protection and the associated certified noise and/or emissions are unchanged and remain applicable to this certificate approval without any interventions.</td>
<td>EASA STC 10076469 (FAA STC ST02091LA - B737-800 Passenger Freighter Conversion) must be concurrently installed. Prior to installation of this change/repair it must be determined that the airworthiness/compatibility is maintained and any previously installed change and/or repair will be removed.</td>
<td>01/05/2021</td>
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<td>ARMSTRONG AEROSPACE Inc. d/b/a ARMSTRONG AEROSPACE</td>
<td>Installation of Shoulder Harness Kit - SH99-220 Revision 1: Models Bell 412 and 412EP added. EASA Certification Basis Updated</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any intervention.</td>
<td>Flight Manual Supplement FMS/1017-043 Issue AT-PED Compatibility Report DR/1017-004 Issue 191M-99-001-MDL, Master Data List, Doc. No. DC10-9001-00, Rev. N, dated 7 May 2023: FAA approved Airplane Flight Manual Supplement as defined in AIR-AMM-1001, Revision A, dated 11 December 2021. Prior to installation of this change/repair it must be determined that the airworthiness of the aircraft remains maintained and any previously installed change and/or repair will be removed. Prior to installation of this airworthiness/compatibility is maintained and any previously installed change and/or repair will be removed.</td>
<td>25/07/2023</td>
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<td>SPECIALIST AVIATION SERVICESLIMITED</td>
<td>Installation of Shoulder Harness Kit - SH99-220 Revision 1: Models Bell 412 and 412EP added. EASA Certification Basis Updated</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any intervention.</td>
<td>Flight Manual Supplement FMS/1017-043 Issue AT-PED Compatibility Report DR/1017-004 Issue 191M-99-001-MDL, Master Data List, Doc. No. DC10-9001-00, Rev. N, dated 7 May 2023: FAA approved Airplane Flight Manual Supplement as defined in AIR-AMM-1001, Revision A, dated 11 December 2021. Prior to installation of this change/repair it must be determined that the airworthiness of the aircraft remains maintained and any previously installed change and/or repair will be removed. Prior to installation of this airworthiness/compatibility is maintained and any previously installed change and/or repair will be removed.</td>
<td>25/07/2023</td>
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<td>ALPINE AEROTECH LP</td>
<td>Installation of Shoulder Harness Kit - SH99-220 Revision 1: Models Bell 412 and 412EP added. EASA Certification Basis Updated</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any intervention.</td>
<td>Flight Manual Supplement FMS/1017-043 Issue AT-PED Compatibility Report DR/1017-004 Issue 191M-99-001-MDL, Master Data List, Doc. No. DC10-9001-00, Rev. N, dated 7 May 2023: FAA approved Airplane Flight Manual Supplement as defined in AIR-AMM-1001, Revision A, dated 11 December 2021. Prior to installation of this change/repair it must be determined that the airworthiness of the aircraft remains maintained and any previously installed change and/or repair will be removed. Prior to installation of this airworthiness/compatibility is maintained and any previously installed change and/or repair will be removed.</td>
<td>25/07/2023</td>
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<td>GLOBAL AEROSPACE DESIGN CORPORATION</td>
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<td>ST04434CH</td>
<td>A321-253N A321-211/-212/-213N A321-211/-212/-213/-214/-215/-216 A320-214/-233 A320-214/-233 A319-132 A319-115/132 A319-115</td>
<td>Installation &amp; Activation of the ARINC 791 Ka-Band SATCOM System Rev. 1 corrects the Technical Documentation and Limitations/Conditions.</td>
<td>- Global Aeronautic Design Corp. Master Drawing List 18GR146-MLD, dated 16 October 2019, or later FAA approved revision. - Instructions for Continued Airworthiness 18GR146-ICA, Revision B dated 11 December 2019, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>CARLISLE INTERCONNECT TECHNOLOGIES</td>
<td>REV. 1</td>
<td>ST04334CH</td>
<td>SF 25 C</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. - Equivalent Safety Finding(s): EASA 0510564410-001 CR C/Q VIBRATION BUFFETING. The requirements for environmental protection and the associated certified noise and/or emission levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List, Document No. 900-102529 Revision HAFM Supplement 900-102531 Revision Instructions for Continued Airworthiness for Configurations 1-5 Document No. 900-102532 Revision Instructions for Continued Airworthiness for Configurations 6-10 Document No. 900-106369 Revision AEWIS ICA Source Doc...</td>
<td>Airplane Flight Manual Supplement, Document No. 900-102531 Revision Bstructions for Continued Airworthiness (ICA) for Configurations 1-5 Document No. 900-102532 Revision Bstructions for Continued Airworthiness (ICA) for Configurations 6-10 Document No. 900-106369 Revision A. Prior to installation...</td>
<td>11/01/2021</td>
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<td>MOTORSEGLER-CLUB BERLIN e.V.</td>
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<td>BD-700-1A11 BD-700-1A10</td>
<td>Installation of Iridium Satcom Antenna Mirored Location of the 1st Iridium Antenna Installation (FS 767 and 781) in the skin panel from A/C CL (STRO) RHI to the LHS. Extension of approved Model BD-700-1A10 to include BD-700-1A11.</td>
<td>Only for SF 25 C Falka, D-COKX, SN 41436. Prior to installation of this design change it must be determined whether the interrelationship between this design change and any other previously installed design change/repair will introduce an adverse effect upon the airworthiness of the product.</td>
<td>18/12/2020</td>
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<td>ALAMO ENGINEERING GmbH</td>
<td>REV. 1</td>
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<td>A321-271N/-272N A321-251N/-252N/-253N A321-211/-212/-213/-214/-215/-216 A320-211/-212/-213/-214/-215/-216 A319-151N/-152N/-153N A319-151N/-152N/-153N/-154N/-115</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emission levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>- MOC/015-1A5E-01 Rev. C-ICAO/01A5E-01 Rev. Cor later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The STC only covers the structural integration of the Iridium antenna. Any function and operation of the antenna will be subject to another engineering approval. Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previous change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/03/2022</td>
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<td>LUFTHANSA TECHNIK AG</td>
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<td>A321-271N/-272N A321-251N/-252N/-253N A321-211/-212/-213/-214/-215/-216 A320-211/-212/-213/-214/-215/-216 A319-151N/-152N/-153N A319-151N/-152N/-153N/-154N/-115</td>
<td>Exchange of emergency exits slides Rev. 1: Correction of models</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emission levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>Description of Change BD/01-25 Revision 5(Supplemental Type Design Document STD/225-2003M Issue 0 Rev 0) Operational Requirements Document BD/01-25-DR-001 Rev 0 Instructions for Continued Airworthiness A-05-2047-ICA Issue 1 Rev 0 or later revisions of the above listed document(s) approved/accepted...</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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### List of EASA Supplemental Type Certificates

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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<th>Status</th>
<th>System Record</th>
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**10075271**

**STC Holder:** MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)

| Aircraft | Certification Basis for the original product as amendedallengments or alternative airworthiness requirements. -The following paragraphs(s) at a later amendment: C-23 AMEND Initial Issue C-23.1038 and C-23.1038 Amendment 4: C-23 AMEND Initial Issue C-23.1038 and C-23.1038 Amendment 2: The requirements for environmental protection. | Installed and maintained in accordance with Medavia Master Document List Ref. MODE-0310_MDL Revision 1 dated 18/12/2020 operated in accordance with the Medavia Flight Manual Supplement Ref. MODE-0310_AFM-01 Revision 1 dated 02/09/2020. | Due to the single NVS TARA indicator installation on the left-hand side, the operation of the TCAS 7.1 system as certified under the design change, is restricted to single pilot operations. ONLY Prior to installation of this change/repair it must be determined that there is an acceptable relationship between this change/repair and any other previously installed change and/or repair in... | 18/12/2020 | Active | 1077271 |

**10075272**

**STC Holder:** GROB AIRCRAFT SE

| Aircraft | Certification Basis for the original product as amendedallengments or alternative airworthiness requirements. -The following paragraphs(s) at a later amendment: Paragraphs affected by this change as per Certification Program C-208A-02 issue 2 F-AR23 regulation as a... | Master Certification Document List, C-208A, issue 2, dated 26 October 2020. To be installed according to Aircraft Maintenance Manual Supplement, C-208A-AMM-01, dated 21 October 2020. To be operated a... | External equipment to be mounted on the wing hard points (not approved by this STC approval). Limited to S/N: 208B540395. Prior to installation of this change/repair it must be determined that there is an acceptable relationship between this change/repair and any other previously installed change and/or repair in... | 24/04/2021 | Active | 1077272 |

**10075283**

**STC Holder:** JET AVIATION AG

| Aircraft | Certification Basis for the original product as amendedallengments or alternative airworthiness requirements. -The following paragraphs(s) at a later amendment: CS-23 Amendment 1.25.25, 25.25, 25.29, 25.29, 25.29, 25.29, 25.29, 25.301, 25.302, 25.305, 25.307, 25.321, 25.337, 25.341, 25.365, 25.366... | JBCSC-M-00199MLD-01 ISSUE 05 MASTER DATA LIST JBCSC-M-00199MLD-01 ISSUE 05 AIRPLANE FLIGHT MANUAL SUPPLEMENT JBCSC-M-00199MLD-01 ISSUE 01 INSTRUCTIONS FOR CONTINUED AIRWORTHINESS JBCSC-M-00199MLD-01 ISSUE 02 INSTRUCTIONS FOR CONTINUED AIRWORTHINESS JBCSC-M-00199MLD-01 ISSUE 06 CERTIFICATION COMPLIANCE SHEET... | Private Operation (not for hire, non-commercial) prior to installation of this change/repair it must be determined that there is an acceptable relationship between this change/repair and any other previously installed change and/or repair and introduction of adverse effect upon the airworthiness of... | 01/03/2021 | Active | 1077285 |

**10075287**

**STC Holder:** PARC AVIATION ENGINEERING SERVICES LTD

| Aircraft | Certification Basis for the original product as amendedallengments or alternative airworthiness requirements. -Special Condition (i): E-4105: Type III Overture Emergency Exit Access Group 1 & E-2105: Type III Overture Emergency Exit Access Group 2: -The following paragraph(s) a... | Master Document List 320-25317-MDL rev 00 dated 22.12.2020 Aircraft Flight Manual Supplement 320-25317-AMF-01 rev 00 (A319) 320-25317-AMF-02 rev 00 (A320) 320-25317-AMF-04 rev 00 (A321) Master Document List... | As stated in AIRM Supplement. Prior to installation of this change/repair it must be determined that there is an acceptable relationship between this change/repair and any other previously installed change and/or repair and introduction of adverse effect upon the airworthiness of the product. | 21/12/2020 | Active | 1077287 |

**10075293**

**STC Holder:** AEROSPACE DESIGN & COMPLIANCE LLC

<p>| Aircraft | Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise level or emissions levels of the product are unchanged. | Installed in accordance with Aerospace Design &amp; Compliance LLC Master Data List (MDL) ref P1885-MDL-006 REV IR dated September 23, 2020. Operate in accordance with the Aerospace Design &amp; Compliance LLC Aeronautical Flight Manual Supplement (AFMS) ref P1885-AFMS-006 Rev IR dated September 23, 2020. | This STC is applicable to all... | 21/12/2020 | Active | 1077293 |</p>
<table>
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<tr>
<th>STC Holder</th>
<th>BABCOCK</th>
<th>STC Ref</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10075304</td>
<td>APTOZ EHF</td>
<td>767-200</td>
<td>REV 1</td>
<td>767-300</td>
<td>Transportation of cargo on the floor in passenger compartmentRev. 1 extends the calendar limitation to 31 July 2022.</td>
<td>The Certification Basis for the original product as amended by subsequent additional or alternative airworthiness requirements - Deviation(s): DEV 0-60073418 &quot;Transportation of cargo in passenger compartments&quot;. The following paragraph(s) at a later amendment: JAP 25.730 at change B/JAR 25.858 at ch...</td>
<td>- Master Data Unit MDL251156 Rev 0, dated 7 Oct 2021 - Aircraft Flight Manual Supplement AFM251156-1 Rev E, dated 7 Oct 2021 - Instructions for Continued Airworthiness ICA151156-1 Rev D, dated 7 Oct 2021 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The allowance to transport cargo in passenger compartment is limited to 3000 FH following installation of the change onto 31 July 2022. Whatever occurs first. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair...</td>
<td>19/11/2021</td>
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<td>10075306</td>
<td>BABCOCK CRITICAL SERVICESDESIGN N AND COMPLETIONS LIMITED</td>
<td>EC135 T3</td>
<td>EC135 T2+</td>
<td>EC135 T2</td>
<td>Upgrade of Garmin GTN750 installation enabling Satellite Based Augmentation System (SBAS) Performance Based Navigation (PBN) in accordance with Modification Leaflet and Certificate of design MDL/175-003</td>
<td>The Certification Basis for the original product as amended by subsequent additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: CS-27 Amendment 6 for 27.21(a)(27.301(a),(b),x)(27.321)(27.337)(27.561(a),(b) for 27.601)(27.603). - Rotorcraft Flight Manual Supplement FMS/175-012 Issue 1 - Instructions for Continued Airworthiness ICA175-011 Issue 1 - Modification Leaflet and Certificate of design MDL/175-003 Issue 1 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Limited to helicopters with CPDS codex variant Dual Garmin GTN750 installation as per EASA STC 10059823 Rev. 1 (BABCOCK/MOD/128) and TPED tolerance certification as per EASA STC 10062544 Rev. 2 (BABCOCK/MOD/134) are prerequisite prior to installation of this change/repair it must be determined that...</td>
<td>23/12/2020</td>
<td>Active</td>
<td>10075306</td>
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<tr>
<td>10075315</td>
<td>BABCOCK CRITICAL SERVICESDESIGN N AND COMPLETIONS LIMITED</td>
<td>EC135 T3</td>
<td>EC135 T2+</td>
<td>EC135 T2</td>
<td>Upgrade of Garmin GTN750 installation enabling Satellite Based Augmentation System (SBAS) Performance Based Navigation (PBN) in accordance with Modification Leaflet and Certificate of design MDL/188-014</td>
<td>The Certification Basis for the original product as amended by subsequent additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: CS-27 Amendment 6 for 27.107.21(a)(27.201), 27.301(a),(b), 27.303(27.305)(27.307) 27.321(27.337)(27.561(a),(b),x) for 27.601(27.603). - Rotorcraft Flight Manual Supplement FMS/188-012 Issue 3 - Instructions for Continued Airworthiness ICA188-010 Issue 2 - Modification Leaflet and Certificate of design MDL/188-014 Issue 1 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Limited to helicopters with CPDS codex variant Dual Garmin GTN750 installation as per EASA STC 10059823 Rev. 1 (BABCOCK/MOD/128) is prerequisite prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair...</td>
<td>23/12/2020</td>
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<td>10075326</td>
<td>AIRBUS S.A.S.</td>
<td>A350-941</td>
<td></td>
<td></td>
<td>A350 DAL C++ Cabin Reconfiguration This modification covers the cabin reconfiguration of seating configuration between door 2 and 3, including seat change and seatlocation, to match with future DAL Inlet configuration S The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged/remain applicable to this certificate/ approval without any...</td>
<td>Certification Plan Ref A35-AJA-2003 - Issue 3MA3 A380-164 Issue 1 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/01/2021</td>
<td>Active</td>
<td>10075326</td>
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<tr>
<td>10075330</td>
<td>ACORN WELDING Ltd.</td>
<td>TCCA</td>
<td>SA20-28</td>
<td></td>
<td>DHC-6 SERIES 400 DHC-6 SERIES 300, 310 AND 320 DHC-6 SERIES 200 AND 210 DHC-6 SERIES 100 AND 110 Installation of Acorn Welding 0821107-02 Exhaust Stacks incompliance with Transport Canada Civil Aviation (TCCA) STC STC20-28. The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged/remain applicable to this certificate/ approval without any...</td>
<td>Acorn Welding Engineering Order ED-88-01 Rev 0, March 18, 2020 Acorn Welding Drawing CCS3-040-08-01 Rev 0, March 3, 2020 Acorn Welding Instructions for Continued Airworthiness ICA08-01 Rev. 6, June 22, 2018 for later revisions of the above listed document(s) approved/accepted on behalf of EASA in acc...</td>
<td>Only applicable for DHC-6 Series with PT6A-27 and PT6A-34 engines installed. This STC is not compatible with PTSA-20 engines. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effects upon the airworthiness of the product...</td>
<td>19/01/2022</td>
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<td>10075340</td>
<td>TURK HAVA YOLLARI TEKNIK A.S. (TURKISH TECHNIC INC.)</td>
<td>737-800</td>
<td></td>
<td></td>
<td>C2048-BET737-C48N REFURBISHMENT Replacement of all Weber tour class seats with TSI ELASA-6 aircondition seats reconfiguration of PSLU, LVP modification and renewal of curtains and carpets. The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged/remain applicable to this certificate/ approval without any...</td>
<td>Certification Programme document No CS048-CERP rev. 6 (whichever includes Master Document List(s) and/ or later revisions of the above listed document(s) accepted/approved under the EASA system.</td>
<td>This STC constitutes an Alternative Method of Compliance (AMOC) to the LVP modification requirement (6EASA adopted FAA AD 2019-03-29). All provisions of FAA AD 2019-03-29 that are not referenced below remain fully applicable and are to be complied with accordingly Prior to installation...</td>
<td>07/01/2021</td>
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<td>Models</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA STC SA02726LA</td>
<td>B300C</td>
<td>8300</td>
<td>B300</td>
<td>Replacement of the original Pratt and Whitney PT6A-65A engines and accessories with Pratt and Whitney PT6A-67A engines and MT PropellerTV-27-1 N-C-F-RP</td>
<td>1. Installation 201806-001, Rev.IR, dated 9 January 2020, or later FAA approved revision.2. Airplane Flight Manual Supplement AFMS 201806-07, Rev. IR, dated 3 June 2020, or later FAA Approved revision.3. Instructions for Continued Airworthiness, Document Number ICA201806-30, Rev. IR, dated 9 January 2020, or later FAA Approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>01/01/2021</td>
<td>Active</td>
<td>10075345</td>
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<td>BELL TEXTRON, Inc.</td>
<td>FAA SR11247DS</td>
<td>429</td>
<td></td>
<td></td>
<td>Installation of Connected Aircraft Interface Residing On-board (CAIRO) System</td>
<td></td>
<td></td>
<td>This STC is applicable to all serial numbers for the Bell Textron Canada Limited.</td>
<td>11/01/2021</td>
<td>Active</td>
<td>10075346</td>
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<td>FLAIR AIRLINES, Ltd.</td>
<td>TCAS C-LSA11-115D</td>
<td>ISSUE 4</td>
<td>737-800</td>
<td>Installation of Skyview SAT-200 Satcom System Installation of a low-profile dual GPS/Inertial antenna on the top fuselage/RH of the centreline with an aluminium doubler reinforcement, an SAT-200 transceiver installed in the avionics rack, and two control panels (CDP-300 and DVI-300).</td>
<td></td>
<td></td>
<td>As per Mandatory Maintenance Manual Supplement, ACS-NAI Report No.H176070. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>07/01/2021</td>
<td>Active</td>
<td>10075355</td>
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<tr>
<td>GLOBAL AEROSPACE DESIGNCORPORATION</td>
<td>FAA STC04206CH</td>
<td>737-800</td>
<td>Certification of Automatic dependent Surveillance - Broadcast (ADS-B) OUT System utilizing upgraded Aviation Communication &amp; Surveillance Systems (ACSS) XS-950 Mode-S Transponders paired with the aircraft existing GPS Source or newly installed ADS-B dedicated GPS source Rockwell Collins GLU-920 and one...</td>
<td></td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td></td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>25/02/2021</td>
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<td>10075366</td>
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<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC ST04271CH</td>
<td>BD-100-1A10</td>
<td>Installation of an AVANCE L5 System</td>
<td></td>
<td></td>
<td>1. Approved Flight Manual Supplement, Duncan Aviation, Inc. Doc. No.CERTDOC-1044271, Revision A, approved 1 February 2018 or later FAA Approved revision.2. Instructions for Continued Airworthiness - Airworthiness Limitations, Section 15.0 of Duncan Aviation, Inc. Doc. No. CERTDOC-1044330, RevisionA, app...</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>13/01/2021</td>
<td>Active</td>
<td>10075387</td>
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List of EASA Supplemental Type Certificates

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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>10075392</td>
<td>FOKKER TECHNIEK B.V.</td>
<td>737-400</td>
<td>VHF Antenna Structural Provisions This project certifies the structural provisions for the installation of a VHF antenna on top of the fuselage. Activation of the system is notified by the STC.</td>
<td>The Certification Basis (CB) for the original product remains applicable if this certificate is approved. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged.</td>
<td>10028171) Rev. 1: demonstration of compliance of the Emergency Flotation System. (EASA STC)</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/01/2021</td>
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<td>10075392</td>
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<td>10075425</td>
<td>AEROSPACE DESIGN &amp; COMPLIANCE LLC</td>
<td>FAA STC ST04198NY</td>
<td>FAA APPROVED ML AML25-AML-0073</td>
<td>Installation of Becker Avionics DVC6500 Digital Voice Communication System.</td>
<td>Installation of Becker Avionics DVC6500 Digital Voice Communication System.</td>
<td>Appendix 1: Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25. Amptd: CS 25.301(a)(b), CS 25.303, CS 25.3056(a), CS 25.307(a)(b), CS 25.3141, CS 25.601, CS 25.61, CS 25.62, CS 25.63.</td>
<td>Master Data List AML25-AMM-0073, Rev. A: FAA STC SR0...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/01/2021</td>
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<td>10075425</td>
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<td>10075430</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td>400A 400</td>
<td>Installation of dual Garmin GTX 3000 with single Garmin GPSX3000 and Garmin GA 36 GPS Antenna.</td>
<td>Installation of dual Garmin GTX 3000 with single Garmin GPSX3000 and Garmin GA 36 GPS Antenna.</td>
<td>Installed in accordance with ASR-2020-016-IF1-10-00 dated 14/10/2020: maintained in accordance with ASR-2020-016-IF1-11-00 dated 07/01/2021 and operated in accordance with ASR-2020-016-FMS-01-01 dated 03/11/2020 (MODEL 400), ASR-2020-016-FMS-02-00 dated 03/11/2020 (MODEL 40A) or later revision of I...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/01/2021</td>
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<td>10075433</td>
<td>ECMS AVIATION SYSTEMS GMBH</td>
<td>AS 350 D AS 350 BB AS 350 B AS 350 BA AS 350 BB AS 350 BA AS 350 B</td>
<td>STP01 - SKYTEK Pod The SKYTEK Pod is an external box which is located below the heli-copter fuselage and which is attached to the original cargo hook attachment points. It is equipped with specific sensing and measuring equipment.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: For AS 350 B, AS 350 B2, AS 350 BA, AS 350 B, AS 350 D: CS 27 Initial Issue: CS 27.11, CS 27.21, CS 27.25, CS 27...</td>
<td>Flight Manual Supplement: Initial Issue R0 dated 11th December 2020: Airworthiness Limitations Section, Section 11 of Component Maintenance Manual, Maintenance Manual for STP01 on AS 350, Revision R0 dated 12th January 2021: Master Document List, STP01_MD_EASA, Revision R1 dated 13th January 2021- C...</td>
<td>The installation of the STP01 is only approved in combination with the original skid landing gear. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>20/01/2021</td>
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<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>Date of Issue</td>
<td>Status</td>
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<td>BABCOCK MISSION CRITICAL SERVICES FMANAGEMENT S.A.U.</td>
<td>10075443</td>
<td>ST04272AT-D</td>
<td>F 406</td>
<td>Installation of an MTV-27-1-6-C-F-R(P)/CFR220-55( ) on ASI Aviation (Reims)</td>
<td>Installation Instruction Doc. No. E-3589 Rev. 0To be maintained i.e. MT-Propeller ICA Document No. E-3589 Rev. 0for later revisions</td>
<td>According limitation section of AFM-SP Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>16/02/2021</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>10075450</td>
<td>ST04478CH</td>
<td>FAA</td>
<td>Installation of an XM Satellite Weather Receiver and Antenna.</td>
<td>The Certification Basis for the original product becomes applicable to this certificate/approval without any im...</td>
<td>- Gulfstream Aerospace Corporation Index List GA442530601, Revision J, dated 03 November 2012 - Aircraft Flight Manual Supplement, GA41325A001, Revision J, dated 24 October 2012 - Instruction for Continued Airworthiness GA41325A002, Revision B, dated 28 October 2011 for later revisions of the above</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>22/01/2021</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10075462</td>
<td>ST04266AT-D</td>
<td>F 406</td>
<td>Installation of an Executive Cabin Interior ( ) on ASI Aviation (Reims)</td>
<td>The Certification Basis for the original product becomes applicable to this certificate/approval without any im...</td>
<td>- Gulfstream Aerospace Corporation Index List GA442530601, Revision J, dated 03 November 2012 - Aircraft Flight Manual Supplement, GA41325A001, Revision J, dated 24 October 2012 - Instruction for Continued Airworthiness GA41325A002, Revision B, dated 28 October 2011 for later revisions of the above</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>22/01/2021</td>
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<td>MT-PROPPELLER ENTWICKLUNG GmbH</td>
<td>10075453</td>
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<td>F 406</td>
<td>Installation of a Collins Ku-SAT-2000 Tail Mounted SATCOM SystemRev. 1.</td>
<td>The Certification Basis for the original product becomes applicable to this certificate/approval without any im...</td>
<td>- Gulfstream Aerospace Corporation Index List GA442530601, Revision J, dated 03 November 2012 - Aircraft Flight Manual Supplement, GA41325A001, Revision J, dated 24 October 2012 - Instruction for Continued Airworthiness GA41325A002, Revision B, dated 28 October 2011 for later revisions of the above</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>16/02/2021</td>
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<td>WESTERN JET AVIATION, Inc.</td>
<td>10075454</td>
<td>GIV-X</td>
<td>FAA</td>
<td>Installation of a Collins Ku-SAT-2000 Tail Mounted SATCOM SystemRev. 1.</td>
<td>The Certification Basis for the original product becomes applicable to this certificate/approval without any im...</td>
<td>- Gulfstream Aerospace Corporation Index List GA442530601, Revision J, dated 03 November 2012 - Aircraft Flight Manual Supplement, GA41325A001, Revision J, dated 24 October 2012 - Instruction for Continued Airworthiness GA41325A002, Revision B, dated 28 October 2011 for later revisions of the above</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>03/06/2022</td>
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The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated identified noise and/or emissions levels of the product remain applicable to this certificate/ approval without any change.

Prior to installation of this change, repair or modification, the associated noise and/or emissions levels of the product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated identified noise and/or emissions levels of the product remain applicable to this certificate/ approval without any change.

The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated identified noise and/or emissions levels of the product remain applicable to this certificate/ approval without any change.

Prior to installation of this change, repair or modification, the associated noise and/or emissions levels of the product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated identified noise and/or emissions levels of the product remain applicable to this certificate/ approval without any change.

You cannot use descriptive data about this design change duplicated products. This approval is limited to only the installation noted on the FAA STC.

Prior to installation of this change, repair or modification, the associated noise and/or emissions levels of the product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated identified noise and/or emissions levels of the product remain applicable to this certificate/ approval without any change.

Prior to installation of this change, repair or modification, the associated noise and/or emissions levels of the product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated identified noise and/or emissions levels of the product remain applicable to this certificate/ approval without any change.

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<th>System Record</th>
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<tr>
<td>DUNCAHN AVIATION, INC.</td>
<td>Installation of Gogo Biz AVANCE LS Broadband System</td>
<td>The Certification Basis for the original product remains applicable to the certificate approval, except where amended by additional or later versions if indicated on FAA STC ST01854W. The requirements for environmental protection and associated certified noise and/or emissions levels of the system.</td>
<td>1. Duncan Aviation Master Document List, Doc. No. CERTDOC-004745(&gt;)-REV. A, dated 6 October 2017; or later FAA approved revision to Duncan Aviation STC Flight Manual Supplement, Doc. No. CERTDOC-004745, Rev. A, dated 6 October 2017, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair, it must be determined that there will be no adverse effect on the airworthiness of the product.</td>
<td>27/01/2021</td>
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<td>10075490</td>
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<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>CMA-625 SLS Control Panels Replacement</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: CS29.71(a), CS29.71(b), CS29.80(a), CS29.80(b), CS29.83(a), CS29.83(b), CS29.84(a), CS29.84(b), CS29.85(a), and CS29.85(b).</td>
<td>MASTER DOCUMENT LIST - 41244-RPT02 Issue 4 dated 27 January 2021; or later FAA approved revision.</td>
<td>The new approved design changes are only applicable to airplanes previously modified according to CS-25 Amendment 3. Prior to installation of this change/repair, it must be determined that there will be no adverse effect on the airworthiness of the product.</td>
<td>27/01/2021</td>
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<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>External Cargo Basket Installation</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: CS29.71(a), CS29.71(b), CS29.80(a), CS29.80(b), CS29.83(a), CS29.83(b), CS29.84(a), CS29.84(b), CS29.85(a), and CS29.85(b).</td>
<td>MASTER DOCUMENT LIST - 41244-RPT02 Issue 4 dated 27 January 2021; or later FAA approved revision.</td>
<td>The new approved design changes are only applicable to airplanes previously modified according to CS-25 Amendment 3. Prior to installation of this change/repair, it must be determined that there will be no adverse effect on the airworthiness of the product.</td>
<td>27/01/2021</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>A321-200 - CABIN RECONFIGURATION FROM 168C + 1S31YC TO 214Y/C</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: CS29.71(a), CS29.71(b), CS29.80(a), CS29.80(b), CS29.83(a), CS29.83(b), CS29.84(a), CS29.84(b), CS29.85(a), and CS29.85(b).</td>
<td>MASTER DOCUMENT LIST - 41244-RPT02 Issue 4 dated 27 January 2021; or later FAA approved revision.</td>
<td>The new approved design changes are only applicable to airplanes previously modified according to CS-25 Amendment 3. Prior to installation of this change/repair, it must be determined that there will be no adverse effect on the airworthiness of the product.</td>
<td>27/01/2021</td>
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**Notes:**
- **EASA STC no.:** 10075510
- **EASA STC ref.:** ATR 42-200, ATR 42-300, ATR 42-320, ATR 42-400, ATR 42-500, ATR 72-101, ATR 72-201, ATR 72-202, ATR 72-102, ATR 42-500, ATR 42-320, ATR 42-300, ATR 42-200, ATR 42-300
- **Models:** ATR 72-212A, ATR 72-211, ATR 72-212, ATR 72-201, ATR 72-202, ATR 72-101, ATR 72-102, ATR 42-500, ATR 42-320, ATR 42-300, ATR 42-200, ATR 42-300
- **Pressure Altitude:** 25,305, 25,317, 25,321, 25,337, 25,341, 25,361...

**References:**
- easa.europa.eu
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<tr>
<th>No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA</th>
<th>Model</th>
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<th>Certification basis</th>
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<th>Date of Issue</th>
<th>Status</th>
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<td>1007538</td>
<td>URBTE AERO GmbH</td>
<td></td>
<td></td>
<td>HK-36 TTC</td>
<td>MTOM Increase from 770kg to 800kg</td>
<td>Installed in accordance with Engineering Order ref: CDEP77E0v1 dated 28/01/2021</td>
<td>Operated in accordance with Flight Manual Supplement ref: AMF-245-01 rev dated 21/01/2021 or later revisions of the above listed document(s) approved/accept the EASA system.</td>
<td>Tail unit (Metal Tube Type 2, P/N B20-3721-00-020) must be installed/limited to SN 36,541, 36,810 and 36,814. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/01/2021</td>
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<td>1007546</td>
<td>ALAMO ENGINEERING GmbH</td>
<td></td>
<td></td>
<td>EMB-135BJ</td>
<td>INSTALLATION AND ACTIVATION OF IRRIDIUM SATELLITE SYSTEM FOR FLIGHT 1A</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (c): SC H-01 Enhanced Airworthiness Requirements Programme for Aerospace Systems - ICAo EWS-CRI-F-4006/1 Security Protection of Aircraft Systems and Networks The following System (s): - MASTER DATA LIST: Doc No. JSSC-M-03007MED-001, Issue H- AIRPLANE FLIGHT MANUAL SUPPLEMENT: Doc No. JSSC-M-03007FAM-001, Issue-C- INSTRUCTIONS FOR CONTINUED AIRWORTHINESS: Doc No. JSSC-M-03007AAM-001, Issue D-OSS CABIN CREW DATA: Doc No. JSSC-M-03007CDC-001, Issue CDR later revisions of the above listed document(s) approved/accept the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2021</td>
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<td>1007547</td>
<td>JET AVIATION AG</td>
<td>REV. 3</td>
<td>A320-214</td>
<td>VIP CABIN INTERIOR COMPLETION</td>
<td>This modification installs a VIP Cabin Interior into an Airbus A320-214. The aircraft will be outfitted to accommodate 4 Cabin Crew and 35 Passengers Rev. 1</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements. The following System (s): - MASTER DATA LIST: Doc No. JSSC-M-03007MED-001, Issue H- AIRPLANE FLIGHT MANUAL SUPPLEMENT: Doc No. JSSC-M-03007FAM-001, Issue-C- INSTRUCTIONS FOR CONTINUED AIRWORTHINESS: Doc No. JSSC-M-03007AAM-001, Issue D-OSS CABIN CREW DATA: Doc No. JSSC-M-03007CDC-001, Issue CDR later revisions of the above listed document(s) approved/accept the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/11/2022</td>
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<td>1007551</td>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>REV. 1</td>
<td>747-300-400/-400F</td>
<td>Installation of Trash Compactors Revision 1: Transfer of STC holder from Safran Cabin Germany GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any modifications.</td>
<td>Master Data List TR40066, Revision N, dated 02 October 2020 or later revisions of the above listed document(s) approved/accept the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/05/2021</td>
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<td>1007552</td>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>REV. 1</td>
<td>747-300-400/-400F</td>
<td>Installation of Trash Compactors within C. P. (Tailor Galley) Numbers M3268A1, M3278A1, M4244A1 and M4566A Revision 1: Transfer of STC holder from Safran Cabin Germany GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any modifications.</td>
<td>Master Data List TR40011, Revision R dated 02 October 2020 or later revisions of the above listed document(s) approved/accept the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>1007558</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST04334AT-D</td>
<td>GV1/ GV60</td>
<td>Installation of a Viessat Ka-Band Broadband High Speed Data (HSD) System on Gulfstream Model GV1 aircraft.</td>
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<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/02/2021</td>
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### List of EASA Supplemental Type Certificates

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<th>FAA STC</th>
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<th>Foreign/NAA STC ref.</th>
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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<td>10075574</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA SR9512D05</td>
<td>AS 350 B2, AS 350 B3</td>
<td>Installation of extended pilot and copilot seat rails.</td>
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<td>Airbus Helicopters, Inc. Master Drawing List (MDL) 530A63-1793 Rev 14 July 2020</td>
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<td>15/02/2021</td>
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<td>10075581</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC ST01863WI</td>
<td>CL-600-2B16 (604 VARIANT), CL-600-2B16 (601-3R VARIANT), CL-600-2B16 (601-3A VARIANT)</td>
<td>Installation of GoGo Business Aviation AVANCE LS Broadband System</td>
<td></td>
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<td>1. Master Document List, Duncan Aviation, Inc. Doc. No. CERTDOC-004903, Rev. A dated 21 December 2017, or later FAA Approved revisions. 2. FAA STC Document, Duncan Aviation, Inc. Doc. No. CERTDOC-004900, Rev. A dated 21 December 2017, or later FAA Approved revisions.</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the anticipated relationship between this change/repair and any other pre-existing change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/02/2021</td>
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<td>10075584</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC ST01955WI</td>
<td>BD-100-1A10</td>
<td>Installation of Aviation Clean Air Bi-Polarization System</td>
<td></td>
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<td>1. Duncan Aviation, Inc. STC Master Document List Doc. No. CERTDOC-011389, Rev. A, dated 30 October 2020, or later FAA approved Revisions. 2. Duncan Aviation, Inc. Instructions for Continued Airworthiness, Doc.No.CERTDOC-011437, Rev. A, dated 30 October 2020, or later FAA approved revisions.</td>
<td></td>
<td>Prior to installation of this change/repair the interrelationship between this change/repair and any other pre-existing change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/02/2021</td>
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<td>10075588</td>
<td>365 AEROSPACE DDS LIMITED</td>
<td>ATR 72-212 A, ATR 72-212 A, ATR 72-211, ATR 72-212</td>
<td>ATR 72-221</td>
<td>Change to Introduce an Automatic Dependent Surveillance – Broadcast (ADS-B) Out System and Single U-SL Flight Management System including Localizer Performance with Vertical Guidance (LPV) Monitor</td>
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<td>Master Data List MDL-1002 Revision A, Flight Manual Supplement No. 1, FMS-1002-001 Revision B, Flight Manual Supplement No. 2, FMS-1002-002 Revision H, Instructions for Continued Airworthiness, ICA-1002-001 Revision C, Section 4.2 Instructions for Continued Airworthiness, ICA-1002-002 Revision C, Section 4...</td>
<td></td>
<td>For Airworthiness Limitations refer to Instructions for Continued Airworthiness, document: ICA-1002-001 and document: ICA-1002-002 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other pre-existing change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/02/2021</td>
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<td>10075589</td>
<td>SABBENA TECHNICS BGC</td>
<td>A330-200</td>
<td>A330-200 - CABIN &amp; IFE Modification from 326YCC to 248C &amp; 265YC</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - The following paragraph(s) at a later amendment:CS-25 at amendment 4 paragraph 1, 29, 301,303, 355,357, 321, 337, 345, 365, 661, 562, 601, 604, 605, 606, 608, 611, 61...</td>
<td></td>
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<td>Master Data List, Ref. 49684-01-A-MDL-R00 dated 01 Feb 2021(+)(+) &gt;FAA STC Document, Duncan Aviation, Inc. Doc. No. CERTDOC-004900, Rev. A dated 21 December 2017, or later FAA Approved revisions.</td>
<td></td>
<td>As a concurrent requirement of this change, accomplishment of this STC requires the simultaneous accomplishment of the following modification - Installation of a photo-luminescent EPN as per Luftfahrtb Sachtw Technik EASA STC No. 100450 Rev.02 Prior to installation of this change/repair it must be determined...</td>
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<td>EASA STC no.</td>
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<td>10075592</td>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FAA STC ST03606NY</td>
<td></td>
<td>MYSTÈRE FALCON 900 (MF900) FALCON 900EX (F900EX)</td>
<td>Installation of Mid-Continent Instruments and Avionics MD302 StandbyAttitude Module</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST03606NY. The requirements for environmental protection and associated noise and/or emissions levels of the original product remain applicable, except where amended by additional or later amendments if indicated on FAA STC ST03606NY.</td>
<td>Dassault Aircraft Services Master Data List, F9XW-D0095-005, Revision IR, dated 11 September 2020 or later FAA approved revision. 3. Instructions for Continued Airworthiness (ICA) F9XW-D0095-155, Revision IR, dated 3 September 2020, or later FAA approved revision. 4. Airplane Flight Manual Supplement for Dassault Falcon 900</td>
<td>08/02/2021</td>
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<td>10075604</td>
<td>MISSION SYSTEMS DAVENPORT INC.</td>
<td>REV. 2</td>
<td></td>
<td>A320-212/213/214/215/216/217/218</td>
<td>Cobham A320 series Air Separation Module (ASM) EASA STC validation of the FAA PMA - Cobham A320 Air Separation Module (ASM) Part Number 3261217-0101 Rev. 1: Name change from Cobham Mission Systems Davenport LSS Inc. to Mission Systems Davenport Inc. Rev. 2: as per PMA approval PQ42450CE Supplement No.02 rev 1...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: CS 25.301 – amdgt 16, CS 25.305 – amdgt 16, CS 25.307 – amdgt 18, CS 25.307 – amdgt 18, CS 25.616 – amdgt 16, CS 25.601 – amdgt 1...</td>
<td>- Pub. No. 25180A, European Aviation Safety Agency (EASA) - Supplements/Type Certificate (STC), Master Document List (MDL) for PN 3261217-0101 for Airbus A318-100, A320-200, A321-100/200 Pub. No. 29152. Instructions for continued airworthiness (ICA) for the Air Separation Module (ASM) PIN 32...</td>
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<td>GARMIN INTERNATIONAL, Inc.</td>
<td>PA-46-500TP</td>
<td>SAU1947W1</td>
<td>Hardware and software updates for the G1000 Integrated Avionics System</td>
<td>The Certification Basis for the original product remains applicable.</td>
<td>FAA STC 100052498 (Ref: 40696-01-A-ICA Rev 4)</td>
<td>Monogram Systems Top Drawings List MDL7000-2, Revision G, dated 10 August 2020</td>
<td>Prior to installation of this product, it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2021</td>
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<td>10075630</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST04321AT-D</td>
<td>GULFSTREAM G280</td>
<td>Installation of Vaseat GAT-5510 Ke-Board High Speed Data (HSD) System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Gulfstream Aerospace Corporation Index List GA413208002, Revision &quot;<em>&quot;, dated 28 March 2019, Aircraft Flight Manual Supplement, GA413208001, Revision &quot;</em>&quot;, dated 28 March 2019, Instruction of Continued Airworthiness, GA413208003, Revision &quot;*&quot;, dated 17 November 2019 for later revisions of the above listed document(s) as applicable.</td>
<td>The EASA STC 100052498 (FAA ST04321AT-D), Installation of an Executive Passenger Cabin Interior(s) is required as part of this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2021</td>
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<td>10075631</td>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>REV. 2</td>
<td>777-200</td>
<td>Installation of Monogram Systems Chiker PN 7000-2(Revision): Transfer of STC holder from Safran Cabin Galleys US Inc. to Safran Cabin Gehrmany GmbH/Revision 2: Correction of Master Data List</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Monogram Systems Master Data List MDL7000-2, Revision G, dated 10 August 2020</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10075632</td>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>REV. 1</td>
<td>747-400</td>
<td>Installation of Trisha Compactor Revision 1: Transfer of STC holder from Safran Cabin Galleys US Inc. to Safran Cabin Germany GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Monogram Systems Top Drawing List Report Number TR40369, Revision G Guided 10 August 2020</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/05/2021</td>
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<td>10075633</td>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>REV. 1</td>
<td>A330-343</td>
<td>Installation of Trisha Compactor Assembly/Revision 1: Transfer of STC holder from Safran Cabin Galleys US Inc. to Safran Cabin Germany GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Monogram Systems Top Drawing List Report Number TR40572, Revision Guided 10 August 2020</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10075634</td>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>767-300</td>
<td>Installation of Trash Compactor:Revision 1. Transfer of STC holder from Safran Cabin Galleys US Inc. to Safran Cabin Germany GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im…</td>
<td>Top drawing list TR40608, Revision J, dated 10 August 2020&lt;…&gt;or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10075635</td>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>747-400, 777-200</td>
<td>Installation of Trash Compactor Assembly:Revision 1. Transfer of STC holder from Safran Cabin Galleys US Inc. to Safran Cabin Germany GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im…</td>
<td>Top Drawing List TD40389, Revision H, dated 10 August 2020&lt;…&gt;or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2021</td>
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<td>SAFRAN CABIN GERMANY GmbH</td>
<td>747-200C, 747-200F</td>
<td>Installation of Aero-Design Technology trash compactors in C. F. Taylor Galleys M4945-A1 and M4955-A1:Revision 1. Transfer of STC holder from Safran Cabin Galleys US Inc. to Safran Cabin Germany GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im…</td>
<td>AERO-DESIGN TECHNOLOGY Master Drawing List TR40201, Revision J, dated 10 August 2020&lt;…&gt;or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>SAFRAN CABIN GERMANY GmbH</td>
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<td>Installation of Trash Compactor:Revision 1. Transfer of STC holder from Safran Cabin Galleys US Inc. to Safran Cabin Germany GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im…</td>
<td>AERO-DESIGN TECHNOLOGY Incorporated Top Drawing List TR40608, Revision J, dated 10 August 2020&lt;…&gt;or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10075638</td>
<td>SAFRAN CABIN GERMANY GmbH</td>
<td>757-200</td>
<td>Installation of Trash Compactor:Revision 1. Transfer of STC holder from Safran Cabin Galleys US Inc. to Safran Cabin Germany GmbH</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im…</td>
<td>AERO-DESIGN TECHNOLOGY Incorporated Top Drawing List TD20106, Revision K, dated 10 August 2020&lt;…&gt;or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2021</td>
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**List of EASA Supplemental Type Certificates**

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<th>STC no.</th>
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
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<tr>
<td>10075639</td>
<td>HELI AUSTRIA GmbH</td>
<td>A412 412</td>
<td>HAU-O-022 Installation of Garmin GTN750 COM/NAV/GPS System and Master Caution Panel Relocation Reason for Rev. 1. addition of A412 to list of approved to replace the No. STC750 by a GNCS2 GNOS/COM/NAV unit, the installation of an additional SSD120 altitude encoder model. Further a new P/N of the GTN750 incorrect</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im…</td>
<td>Master Document List Doc. No. HAU-O-022-001, Rev. 1</td>
<td>For the Bell Model 412 the change limited to serial numbers 33001−36086&lt;…&gt;</td>
<td>25/04/2022</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>FAA STC or SAR No.</td>
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<td>Description</td>
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<td>10075646</td>
<td>BELL TEXTRON, Inc.</td>
<td>FAA STC SR11246DS</td>
<td>505</td>
<td>Installation of Flitestep</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SR11246DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Remains applicable to this certificate/ approval without any impact.</td>
<td>This STC is applicable to all serial numbers for the Bell Textron Canada Limited. Prior to installation of this change/repair it must be determined if the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the...</td>
<td>16/02/2021</td>
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<tr>
<td>10075649</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>EMB-13SLR</td>
<td>EMB-135 Cabin Reconfiguration: This modification consists in switching the cabin dual class version/VP-passengers of Embraer EMB-135LR into passengers stretch configuration. Two medical stretchers with P/N: KIT-AS-18113-175 manufactured by ULD Systems, are installed in the FWD RH cabin section. Stretchers...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Remains applicable to this certificate/ approval without any impact.</td>
<td>Limited to model previously modified with EASA STC 10041778.</td>
<td>19/02/2021</td>
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<tr>
<td>10075659</td>
<td>BELL TEXTRON, Inc.</td>
<td>FAA SR11246DS</td>
<td>407</td>
<td>Installation of Connected Aircraft Interface Residing On-board (CAIRO) System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SR11246DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Remains applicable to this certificate/ approval without any impact.</td>
<td>Applicable to serial numbers 53000 to 54299 with STC EASA 10073820 (FAASR11246DS - 407 G1000NXi Glass Cockpit Upgrade) and to serial numbers 54805 and subsequent. Prior to installation of this change/repair it must be determined if the relationship between this change/repair and any other previous...</td>
<td>17/02/2021</td>
<td>Active</td>
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<tr>
<td>10075680</td>
<td>S-TEC CORPORATION</td>
<td>FAA SA09762DS</td>
<td>PA-24, PA-24-250 PA-24-260 PA-24-400</td>
<td>Installation of an S-TEC G1000 Upgrade Flight Control System (DFCS) Rev. 1 connects the AAF to reference.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA09762DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Remains applicable to this certificate/ approval without any impact.</td>
<td>Prior to installation of this change/repair it must be determined if the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/03/2021</td>
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<tr>
<td>10075661</td>
<td>B.A.S., Inc.</td>
<td>FAA SA08555SE</td>
<td>T210M, T210N, T210R, P210N, P210R, T210K, T210L, 210K, 210L, 210M, 210R</td>
<td>Installation of B.A.S., Inc. dual belt shoulder harness with inertia reel</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged. Remains applicable to this certificate/ approval without any impact.</td>
<td>Prior to installation of this change/repair it must be determined if the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/02/2021</td>
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<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10075664</td>
<td>FTTI ENGINEERING NETWORK GmbH</td>
<td></td>
<td></td>
<td>CL-600-2B19 (RJ SERIES 100)</td>
<td>Introduction of ADSB-out capability</td>
<td>- System Description F30-EO19-0060, Issue 02 - Instructions for Continued Airworthiness F33-EO19-0060, Issue 02</td>
<td>- Maintenance Manual Supplement F21-001-000, Issue 02 - FAA STC ST02829CH</td>
<td>This STC is only for aircraft which have Collins Proline 4 avionics already installed on factory. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/02/2021</td>
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<td>10075667</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 1</td>
<td></td>
<td>A350-941</td>
<td>Installation of non-electrical Floor Path Marking System</td>
<td>Supplemental Type Definition Document STD033-1908M Issue 1 rev.1 - Airplane Flight Manual Supplement A-00-1908-AFM Issue2 rev 0 - Instructions for Continued Airworthiness A-60-1908-AF Issue 1 rev 0</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/03/2021</td>
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<td>10075706</td>
<td>URBE AERO GmbH</td>
<td>REV. 2</td>
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<td>HK 36 TTC HK 36 TC</td>
<td>MTOM increased</td>
<td>For Model HK 36 TTC The change is defined with Revision 1 of Urbe Aero GmbH EngineeringOrder DOE/F07 Change 240 dated 18.02.2021. The aircraft shall be operated in accordance with initial issue of Urbe Aero GmbH Airplane Flight Manual Supplement document AFM-240-01 dated19.12.2020 For Model HK 36 TTC No aircraft</td>
<td>1. Metal Tube Type 2 with Part Number 820-3721-00-02 must be installable to increase of the MTOM. Check of the aircraft by Urbe Aero inaccordance with Doc No. DOE/F07 Change 240 for suitability is a prerequisite for the MTOM increase. Prior to installation of this change/repair it must be determined...</td>
<td>16/12/2022</td>
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<td>10075709</td>
<td>BROMMIA MAINTENANCE AB</td>
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<td>FALCON 7X</td>
<td>Stretcher installation on Falcon 7X</td>
<td>- MDD-20-04 Rev. D - FMS-20-04 Rev. D - ICA-20-04 Rev. Cor later revisions of the above listed document(s) approved/ accepted/under the EASA system.</td>
<td>Patient transport is limited to only using Liftport Aerolift TB&lt;-&gt;&lt;-&gt;HPN: A31-001-01 together with FB2-007 mattress and utility arch P/N: 106-4045. Prior to installation of this change/repair it must be determined that...</td>
<td>23/02/2021</td>
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<td>10075710</td>
<td>BOEING IRELAND LIMITED</td>
<td>REV. 3</td>
<td></td>
<td>737-800</td>
<td>Removal of the Row 44 Wi-Fi antenna installation</td>
<td>Master Data List provided in Change/Repair Approval Document, Doc:BIL-00177-APP Rev.0 (Instructions for Continued Airworthiness, Doc: BIL-00177-ICAO1 Rev. 0) for Continued Airworthiness, Doc: BIL-00177-ICAO2 Rev. 0 for Continued Airworthiness, Doc: BIL-00177-ICAO3 Rev. 0 for Continued Airworthiness, Doc: BIL-00177-ICAO4 Rev. 0</td>
<td>A pre-requisite to this change is that the Row 44 broadband system (EASA/STC 10033186 FAA STC ST03834CH-D) is deactivated and removed. It will be ann...</td>
<td>10/06/2024</td>
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</table>
List of EASA Supplemental Type Certificates

**STC Holder**

1. **S-TEC CORPORATION**
   - FAA STC SA09763DS
   - Revision 1
   - Installed of an S-TEC 01320 Digital Flight Control System (DFCS) Rev. 1
   - Certification basis: Master Data List ST-853 MLD-0001, Revision A, dated 21 January 2020
   - Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.
   - Issue date: 25/02/2021
   - Status: Active
   - System Record: 10075713

2. **COLUMBIA AVIONICS INC.**
   - FAA STC SA01958WI
   - SEE FAA APPROVED MODEL LIST
   - Installation of a Collins KuSAT-2000 Tail-Mounted SATCOM System
   - Certification basis: FAA STC SA01958WI
   - This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List SA01958WI dated 24 January 2020.
   - Issue date: 25/02/2021
   - Status: Active
   - System Record: 10075723

3. **WESTERN JET AVIATION, Inc.**
   - FAA STC ST04860CH
   - SEE FAA APPROVED MODEL LIST
   - Installation of a Collins KuSAT-2000 Tail-Mounted SATCOM system system Rev. 1
   - Certification basis: Western Master Data List 1469990 Revision B dated 13 January 2021
   - This STC requires prior or concurrent embodiment of a TPED STC Service Bulletin. Prior to installation of this change/repair the STC must be determined that the interrelationship between these change/repair and any other previously installed change/repair will introduce no adverse effect upon the product.
   - Issue date: 03/08/2022
   - Status: Active
   - System Record: 10075724

4. **NORTHERN AVIONICS S.R.L.**
   - FAA STC SA00561W
   - WIAS GPS ANTENNA INSTALLATION
   - Certification basis: Western Master Data List 1469990 Revision B dated 13 January 2021
   - This STC requires prior or concurrent embodiment of a TPED STC Service Bulletin. Prior to installation of this change/repair the STC must be determined that the interrelationship between these change/repair and any other previously installed change/repair will introduce no adverse effect upon the product.
   - Issue date: 25/02/2021
   - Status: Active
   - System Record: 10075726

5. **3S CERTIFICATION, LLC**
   - FAA STC SA01856WI
   - SEE FAA APPROVED MODEL LIST
   - Installation of the Send Solutions Airtext Router
   - Certification basis: 1. 3S Certification, Master Data List (MDL) 3817-002-1001, Revision 3.0, dated 13 January 2020
   - This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List SA01856WI dated 13 January 2020.
   - Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the product.
   - Issue date: 25/02/2021
   - Status: Active
   - System Record: 10075728

6. **SAFRAN CABIN CALLEYS USA, INC.**
   - FAA STC ST01916A
   - A320-233
   - Certification basis: Installation of an A3 Stowage Unit PN: 502331-1
   - Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.
   - Issue date: 02/03/2021
   - Status: Active
   - System Record: 10075729

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**Associated Technical Documentation**

- Master Data List ST-853 MLD-0001, Revision A, dated 21 January 2020
- FAA Flight Manual Supplement ST-853-APM0001, Revision 2, dated 22 December 2020
- FAA STC SA09763DS dated 22 December 2020
- FAA STC SA01958WI dated 13 January 2021
- FAA STC SA01856WI dated 13 January 2021
- FAA STC SA01856WI dated 13 January 2021

**Limitations/Conditions**

- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.

**Date of Issue**

- 25/02/2021
- 25/02/2021
- 03/08/2022
- 25/02/2021
- 25/02/2021
- 02/03/2021

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**Revision History**

- Master Data List ST-853 MLD-0001, Revision A, dated 21 January 2020
- FAA Flight Manual Supplement ST-853-APM0001, Revision 2, dated 22 December 2020
- FAA STC SA09763DS dated 22 December 2020
- FAA STC SA01958WI dated 13 January 2021
- FAA STC SA01856WI dated 13 January 2021
- FAA STC SA01856WI dated 13 January 2021

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**List of EASA Supplemental Type Certificates**

- **A320-211**
- **A320-212**
- **A320-213**
- **A320-214**
- **A320-215**
- **A320-216**
- **A320-217**

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**Related Documents**

- FAA Approved Model List
- FAA STC SA01958WI
- FAA STC SA01856WI
- FAA STC SA00561W
- FAA STC SA01856WI
- FAA STC SA00561W

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**Contact Information**

- Driessen Aircraft Interior Systems Master Data List
- FAA Flight Manual Supplement ST-853-APM0001, Revision 2, dated 22 December 2020
- FAA Approved Model List

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**Notes**

- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.
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<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA</th>
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<th>Certification Basis</th>
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<tr>
<td>1007536</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST0957AT-D</td>
<td>G0-SP</td>
<td>Installation of Interior Emergency Lights</td>
<td>Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/02/2021</td>
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<td>1007540</td>
<td>ARMSTRONG AEROSPACE INC.d/b/a ASTRONICS ARMSTRONG AEROSPACE</td>
<td>FAA ST04665CH</td>
<td>A320-231/232/-233 A320-211/212/-214/-216</td>
<td>Removal of a 2Ku In-Flight Entertainment and Connectivity (IFE) System</td>
<td>Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments: CS-25 amdts 0, 25.29(a)(y), 25.30-3, 25.305(y)(x), 25.307-3, 25.366(a)(y)(x)(y), 25.581, 25.571(a)(y)(a), 25.651, 25.653(a)(y)(x), 25.655(a)(y), 25.661, 25.693(a)(y)(x), 25.701, 25.703(a)(y)(x), 25.711(a)(y)(x).</td>
<td>Airplane Flight Manual Supplements, Doc. No. 0664-09891-2407 and Doc. No. 0684-09909-2406 are not applicable to the aircraft modified by this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>1007542</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>421B</td>
<td>DUAL GPS AND RADAR ALTIMETER INSTALLATION</td>
<td>Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>42104-RF140 Master Document List Ed/142104- HDK01 Flight Manual Supplement Ed/142104-HDK02 Maintenance Manual Supplement Edic. 10 revision of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Cessna 421B Airplane with MSN: 421B002A to 421B004A. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/03/2021</td>
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<td>1007542</td>
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<td>1007548</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>GIV</td>
<td>Cabin Modification - The change consists of cabin modifications such as aisle placement/ removal, installation of galley equipment and monumental items, including the replacement of a 3-place divan by two (2) Single seats.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>AFMS JA02-2050 Rev B - Airplane Flight Manual Supplement, dated 26/06/2021 ISA JA02-2050 - Rev A - Instructions for Continued Airworthiness, dated 15/06/2021.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/03/2021</td>
<td>Active</td>
<td>1007548</td>
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<tr>
<td>1007549</td>
<td>AIRBUS S.A.S.</td>
<td>A321-253NX A321-252NX A321-251NX</td>
<td>Airbus Interiors Services MOD. A0112</td>
<td>Airbus Interiors Services MOD. A0112 Cabin Conversion for TITAN/ FOURSEASONS</td>
<td>Certification Basis for the original product as amended. Installation of alternative airworthiness requirements. - The following paragraph at JA02 0111: 25.365, 25.767, 25.785, 25.789, 25.791, 25.815, 25.817, 25.851, 25.1357, 25.1359, 25.1415, 25.1441, 25.1443, 25.1447, 25.1450.</td>
<td>This STC presents the specificity to enable several cabin configurations. Any subsequent modification to the airplane is allowed, provided that the installation between this subsequent design change and each cabin configuration permitted by the STC is evaluated and found to meet the requirement.</td>
<td>02/03/2021</td>
<td>Active</td>
<td>1007549</td>
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<tr>
<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AVIATION Inc.</td>
<td>HAWKER 800XP</td>
<td>HAWKER 850XP</td>
<td>HAWKER 800</td>
<td>HAWKER 700</td>
<td>BAE.125 SERIES 800B</td>
<td>BAE.125 SERIES 800A</td>
<td>BAE.125 SERIES 1000B</td>
<td>BAE.125 SERIES 1000A</td>
<td>ES...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>MYSTÈRE FALCON 50 (MF50)</td>
<td>Upgrade of a Honeywell MK V EGPWS with Optional Runway Safety Package</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. Special Condition(s): ORI-F-8109K001 Upgrade of a Mark V EGPWS Computer with Optional Runway Safety Package. The requirements for environmental protection and the associated certificated noise and/or emissions remain unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Master Data List 1032067 Revision (A), AFM Supplement 1034043 Revision (A), AFM Supplement 1034095 Revision (A), Instructions for Continuous Airworthiness 1033949 Revision (B), MMEL Supplement DF-JAMELS-F50-00-1 Issue 1/1 prior to installation of this change/repair.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/03/2021</td>
<td>Active</td>
<td>10775754</td>
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<tr>
<td>TEXTRON AVIATION Inc.</td>
<td>FAA SA2841CE</td>
<td>1900D</td>
<td>Installation of a Privacy Partition and a non-belted toilet in the cabin</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Drawings and data called out on United Beechcraft, Inc. Top Drawing No.UB1901006, dated 23 July 1992-(x)- (y) prior to installation of this change/repair and any other previously installed change and/or repair.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/03/2021</td>
<td>Active</td>
<td>10775756</td>
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<tr>
<td>TEXTRON AVIATION Inc.</td>
<td>FAA SA2842CE</td>
<td>1900D</td>
<td>Installation of a mid-baggage cargo net, option 1, or an aft bulkhead partition, option 2, in the aft cabin</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Drawings and data called out on Raytheon Aircraft Services Top Drawing No.UB1901005, dated 23 July 1992-(x)- (y) prior to installation of this change/repair and any other previously installed change and/or repair.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/03/2021</td>
<td>Active</td>
<td>10775757</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST02715AT-D</td>
<td>GV-SP</td>
<td>Installation of an External Camera System (Securiplane)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Gulfstream Aerospace Corporation Index List GC51505810, Revision &quot;Y&quot;, dated 07 October 2003. Airplane Flight Manual Supplement GC5150M006, Revision &quot;NC&quot;, dated 07 October 2003. Instructions for Continued Airworthiness, GC515054A005, Revision &quot;B&quot;, dated 03 September 2004. Or later revisions of the above listed document(s) approved/accepted.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/03/2021</td>
<td>Active</td>
<td>10775768</td>
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<td>STC Holder</td>
<td>FAA STC</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>ANORA INTERNATIONAL LLC</td>
<td>ST0275LA</td>
<td>737-800</td>
<td>Installation of a cargo loading system</td>
<td>1. Ancra International LLC Controlled Data List CL68700, Revision E, dated 28 October 2019, or later FAA approved revision 2. Instructions for Continued Airworthiness, report number 10725 Revision A, or later FAA accepted revision.</td>
<td>Installation of IAI's 737-800 &quot;BDSP&quot; Pax to Special Freighter STC (FAA ref ST02625E1) is a pre-requisite for the installation of this STC. EASA validation of this FAA STC is necessary for aircraft to be accepted on the register of a European country. Prior to installation of this change/repair it must be...</td>
<td>09/07/2021</td>
<td>Active</td>
<td>ST0771</td>
<td></td>
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<tr>
<td>SKANDIA, INC</td>
<td>ST04475CH</td>
<td>BD-700-1A10, BD-700-1A11</td>
<td>Installation of Thermal/Acoustic kits</td>
<td>Skanda Master Data List, Document Number 7803-0551ML, Revision A, dated 25 July 2020</td>
<td>Instructions for Continued Airworthiness - Skanda Document No. 7803-0551MCA, No Revision, dated 10 January 2020</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/03/2021</td>
<td>Active</td>
<td>ST0776</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>ST03905NY</td>
<td>MYSTERE FALCON 900 (MF900), FALCON 900EX (F900EX)</td>
<td>EASA Validation of FAA STC ST03905NY/CMU M4 1st upgrade in Dassault Aviation Mystere-Falcon 900 and Falcon600EX with CNUMA III.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval without any im...</td>
<td>Installation in accordance with Master Data List F90XW-D0039-005, Revision 10, dated 25 July 2020</td>
<td>04/03/2021</td>
<td>Active</td>
<td>ST0779</td>
<td></td>
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<tr>
<td>DAHER AEROSPACE/SA DAHER</td>
<td>P.68 &quot;OBSERVER 2&quot;</td>
<td>STC 06-001-19 Installation of flight parameters measuring devices</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval without any im...</td>
<td>Supplemental Type Certificate n ST 06-001-19 Issue 1 dated 12.05.2015; Aircraft Flight Manual Supplement n SUP1_AF_FH_P989SA E_R0 dated 08.12.2020</td>
<td>Installation of this STC is limited to VULCANAIR P.68 &quot;OBSERVER 2&quot; s/n473.</td>
<td>12/03/2021</td>
<td>Active</td>
<td>ST0780</td>
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<tr>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>ST04218CH</td>
<td>MYSTERE FALCON 900 (MF900), FALCON 900EX (F900EX)</td>
<td>Upgrade of a Honeywell MK V EGPWS with Optional Runway Safety Package</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval without any im...</td>
<td>Master Data List 1032081 Revision (E) / (F)</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/03/2021</td>
<td>Active</td>
<td>ST0781</td>
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<td>EASA STC ref.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10075825</td>
<td>JET AVIATION AG</td>
<td>REV 1</td>
<td>787-8</td>
<td>Seat Mounting Intercostal - Provisions Only: This STC introduces metallic intercostal washers which may be embedded inside B787-8/-9 floorboards to allow the installation of seats and cabin furnishing. The STC provides an envelope identifying the allowed configurations in terms of intercostal span, number of washers to be used, and the locations where washers are permitted.</td>
<td>Certification Basis for the original product as amended by the followin...</td>
<td>Master Data List JBS-119-MDL-01, Issu...</td>
<td>Limitations and conditions i.e. B787 Intercostal - Airworthiness Requirements JBS-119-MDL-01, Issue 01. This STC does not include the actual installation of the intercostal washers, B787-8/-9, which is covered via a separate cabin modification STC.</td>
<td>27/09/2021</td>
<td>Active</td>
<td>10075825</td>
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<tr>
<td>10075807</td>
<td>BELL TEXTRON, Inc.</td>
<td>FAA SR11260DS</td>
<td>206L, 206L-1, 206L-3, 206L-4</td>
<td>Installation of Lithium Main Battery Compartment Kit P/N 2561-10180-002A, consisting in a fuselage cut out provision, located within the allowed configurations in terms of intercostal span, number of washers to be used, and the locations where washers are permitted.</td>
<td>Certification Basis for the original product as amended by additional or later amendments if indicated on FAA-STC SR11260DS dated 26 February 2021. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any improvement.</td>
<td>Master Drawing List, Report No. AA-202001, Revision C, dated 11 September 2020</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/03/2021</td>
<td>Active</td>
<td>10075807</td>
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<tr>
<td>10075825</td>
<td>SAPRA...</td>
<td>FAA SR11260DS</td>
<td>206L, 206L-1, 206L-3, 206L-4</td>
<td>Installation of Lithium Main Battery Compartment Kit P/N 2561-10180-002A, consisting in a fuselage cut out provision, located within the allowed configurations in terms of intercostal span, number of washers to be used, and the locations where washers are permitted.</td>
<td>Certification Basis for the original product as amended by additional or later amendments if indicated on FAA-STC SR11260DS dated 26 February 2021. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any improvement.</td>
<td>Master Data List MOD-A35001-MDL, Issue 1, dated 10 Mar 2021</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2021</td>
<td>Active</td>
<td>10075825</td>
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<tr>
<td>10075833</td>
<td>INTERNATIONAL WATER GUARDINDUSTRI EIS Inc.</td>
<td>FAA SR11260DS</td>
<td>206L, 206L-1, 206L-3, 206L-4</td>
<td>Installation of the Potable Water Quantity Pre-select system</td>
<td>Certification Basis for the original product as amended by the followin...</td>
<td>Master Data Summary 1094-MDS-01 Issue 2</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/03/2021</td>
<td>Active</td>
<td>10075833</td>
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<tr>
<td>10075840</td>
<td>PATRIA AVIATION OY</td>
<td>AW119MKII</td>
<td>A350-900 CARICAR Lab installation Rev 0</td>
<td>Installation of G500H EFIS, GTM600COM/NAV and GTX335 TRANSPONDER</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any improvement.</td>
<td>Certification Program D-119-11-CP-01, Revision 8, dated 01.03.2021</td>
<td>Configuration pre-requisites: Rotorcraft configuration is above TCDS-119 and valid EASA type certificate. Rotorcraft configuration is approved/accepted under the EASA, Canada and the US.</td>
<td>11/03/2021</td>
<td>Active</td>
<td>10075840</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10075843</td>
<td>PATRIA AVIATION OY</td>
<td></td>
<td>AVI115MKII</td>
<td>Mission system upgrade</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Certification Program D-119-11-CP-02, Initial Revision, dated 29.01.2021</td>
<td>Configuration Statement D-119-11- CS-02, Initial Revision of Rotorcraft Flight Manual Supplement D-119-11-RFM, Instruction for Continued Airworthiness D-119-11-AIA-01 and Appendix 2</td>
<td>Mission System upgrade</td>
<td>12/03/2021</td>
<td>Active</td>
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<tr>
<td>10075846</td>
<td>FLYHT AEROSPACE SOLUTIONS LTD.</td>
<td>REV. 1</td>
<td>TC-CA SA12-34 ISSUE 2</td>
<td>CL-600-2D04 (RJ SERIES 900) CL-600-2D15 (RJ SERIES 705) CL-600-2C10 (RJ 700, 701, 702) CL-600-2B19 (RJ SERIES 100)</td>
<td>EASA certification of Transport Canada Supplemental Type Certificate (STC) SA12-34 issued to FLYHT 228 Automated Flight Information Reporting System Installation Rev 1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>For STC installation on aircraft models CL-600-2C10, CL-600-2D15 and CL-600-2D24: STC Type design change definition as per FLYHT Aerospace Solutions Ltd., Master Document List121-0070, Rev. 10, and STC Installation must be in accordance with FLYHT Aerospace Solutions Ltd., Installation Inst...</td>
<td>16/03/2021</td>
<td>Active</td>
<td>10075846</td>
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<tr>
<td>10075850</td>
<td>PMV ENGINEERING</td>
<td></td>
<td>747-400, 747-400F</td>
<td>The STC consists of the installation of the ADBB-O System on Boeing 747-400 and 747-400F: Approval configurations are defined in the applicable Aircraft Flight Manual Supplemental reference. AVI-01460-AFMS</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements - The following paragraph(s) at a later amendment: CS-25 amnds 4: 25.130(a)(1)(ii), 25.130(a)(3), 25.130(a)(4), 25.130(a)(6), 25.131(c), 25.130(a)(9), 25.1323(a)(1)(ii), 25.1323(a)(1)(i) and (i)?&gt;25...</td>
<td>Master Document List AVI-01460-MDL-501-R01, dated 29 Sep 2019 - Aircraft Flight Manual Supplement AVI-01460-AFMS-501-R01, dated 16 APR2020 - Instructions for Continued Airworthiness AVI-01460-ICA-501-R00, dated 16 APR 2020: Other previously installed configurations are defined in accordance with FLYHT Aerospace Solutions Ltd., General Document List121-0070, Rev. 10, and STC Installation must be in accordance with FLYHT Aerospace Solutions Ltd., Installation Inst...</td>
<td>STC limitations and approved configurations if defined in the applicable Aircraft Flight Manual Supplemental reference AVI-01460-AFMS: Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of...</td>
<td>12/03/2021</td>
<td>Active</td>
<td>10075850</td>
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<tr>
<td>10075885</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA STO7020SE</td>
<td>737-800</td>
<td>LED Lighting installation</td>
<td>The Certification Basis for the original product remains applicable to those certificates/approval, except where amended by additional Airworthiness: AIR25 amnds 14: 25.869(a)(1)(ii), 25.1303(a)(ii) and 25.1305(a)(ii) and/or 25.1325 amnds 6: 25.1325(c)(1)(i) and 25.1325(c)(1)(ii) and/or 25.1325(c)(1)(ii) and/or 25.1325(c)(1)(ii)</td>
<td>B/E Aerospace, Inc. - FSI (B/E - FSI Master Drawing List (MDL) B/FS/582-001, Revision D, dated 07 March 2021)</td>
<td>Prior or concurrent incorporation of Boeing Service Bulletin 737-25-1857Rev. 00 dated 10 July 2020: In order to be approved and/accepted on behalf of EASA, the technical implementation must be performed in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of...</td>
<td>17/03/2021</td>
<td>Active</td>
<td>10075885</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10075856</td>
<td>UAVIONIX CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SA0041080</td>
<td>SEE AML FAA STC SA0041080</td>
<td>Validation of STC SA0041080: Av-30-C: The Av-30-C is an instrument designed to serve as a primary attitude ordirection indicator, in addition to other functions and installation approvals.</td>
<td>The Certification Basis for the original product as amended by the following paragraph(s) at a later amendment: 23.601, 23.603, 23.671, 23.887, 23.101, 23.103, 23.106, 23.108, 23.109, 23.116, 23.121, 23.123, 23.123, 23.12...</td>
<td>Installation in accordance with Master Drawing List (MDL) UAV-1004-001, Revision C/0.</td>
<td>18/05/2021</td>
<td>Active</td>
<td>10779202</td>
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<tr>
<td>10075862</td>
<td>AVIO DESIGN GROUP INC., dba AVIO DG</td>
<td>REV. 1</td>
<td>EC 130 B4, EC 130 72, AS 350 B, AS 350 B1, AS 350 B2, AS 350 B3</td>
<td>Astronics Max-Viz 1200 or 1400 Sensor Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>Definition of Type Design Change: TCCA approved Avriq-Master Data List2000-00-001 Revision E, dated April 14, 2023. For AS 350 models (1200 or 1400 E/E) Modification Instruction 2000-MI-001, Revision B, dated January 15, 2021. TCCA approved Instructions for Continued Airworthiness 2002-CA-001 Rev...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/06/2023</td>
<td>Active</td>
<td>10779862</td>
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<tr>
<td>10075882</td>
<td>airplus engineering GmbH</td>
<td>T200H</td>
<td>Installation of Skydive Kit</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>Installation description DE-0001-000000, Rev 3, Installation of SkydiveKit Cessna T 206H/Equipment List EO-0001-000000, Rev 0/Drawing Set ZG-0001-000000, Rev 0/Airframe to operator’s manual model Cessna T 206H “Disembarkation of sky divers”. Document No. PO-0001-000001, Rev 4, LBA approved 04 July 200...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/03/2021</td>
<td>Active</td>
<td>10779863</td>
<td></td>
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<tr>
<td>10075883</td>
<td>airplus engineering GmbH</td>
<td>B60</td>
<td>Installation of Garmin VHF Communication Transceiver / VOR/ILS Receiver/GPS Receiver GNS 530, GNS 430 and GTX 330 Modes Transponder</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>Document DE-0008-340000/Flight Manual Supplement Garmin GNS 530, PO-0008-000350, dated 24th of June, 2004/Flight Manual Supplement Garmin GNS 430, PO-0008-000430, dated 24th of June, 2004/Flight Manual Supplement Garmin GTX 330, PO-0008-000350, dated 24th of June, 2004/other...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/03/2021</td>
<td>Active</td>
<td>10779863</td>
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<tr>
<td>10075884</td>
<td>airplus engineering GmbH</td>
<td>DO 228-101</td>
<td>Instrumentation for DLR “FusionProp” Project</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative environmental protection requirements - The following paragraph(s) at a later amendment: Additional FAR 23,Arnlt.1, The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Flight Manual Supplement PO-2225-000000, Issue 0-3/Continuation Manual Supplement MM-2225-000000, Issue 2/Document List MD-2225-000000, Issue 6/Description of Modification DE-2225-000000, Issue 3/Engineering Work Order EW-2225-000000, Issue 3er later revisions of the above listed document(s)...</td>
<td>This STC is only applicable to research-aircraft registered in Germany. The limited timeframe law, Airplus Doc. No MM-2225-000000, 4.2 applicable. The DLR Mod No BS-228-86-10.1 “Elektrische Versorgung, Verwendung der Versuchsaurüstung, Upgrade vom 02.01.2006” needs to be installed Prior to installation of this...</td>
<td>17/03/2021</td>
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<td>10779864</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>ST04316AT-D</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>ST04316AT-D</td>
<td>Installation of Garmin ADS-B In System</td>
<td>The Certification Basis for the original product remains applicable.</td>
<td>Index List CED1260001, Rev. EFIT Manual Supplement CED1260001, Rev. Allocated for continued airworthiness CED1260001, Rev. D0 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/03/2021</td>
<td>Active</td>
<td>10075911</td>
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<tr>
<td>ST04585NY</td>
<td>FAA STC</td>
<td>A91139 AB1139</td>
<td>Replacement of Vertical and Horizontal Stabilizer Lights with LED Position/anti-collision Lights</td>
<td>The Certification Basis for the original product remains applicable this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SR04585NY. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Sterling Helicopter Master Cleaning List. Document No. ST-19-500-006, Rev. A, dated January 19, 2021. Sterling Helicopter Instructions for Continued Airworthiness. Document No. ST-19-500-008, Rev. IR, dated January 7, 2021.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/03/2021</td>
<td>Active</td>
<td>10075923</td>
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<tr>
<td>DA 62</td>
<td>BERINGER AERO</td>
<td>DA 62</td>
<td>Replacement of Thrust Reverser Actuator (TRA)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>MDL-014000/000 Rev. 02, dated March 2021.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/03/2021</td>
<td>Active</td>
<td>10075937</td>
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<td>REV. 1</td>
<td>SPESSART AIR SERVICE GmbH</td>
<td>F152</td>
<td>Installation of Engine Lyncom O-235-L2C into EASA STC is an amendment of former LBA STC (EMZ SA 1217) dated 07.07.2001. The purpose of this revision is only extension of serialnumber applicability.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Flight Manual: &quot;Flight Handbook Reims/CESSNA F152, Ausgabe 2001&quot;, LBA approved 18.07.2001 and/or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The effectiveness of this STC shall be limited to aircraft serial numbers F15201541 and F15201961. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/03/2021</td>
<td>Active</td>
<td>10075972</td>
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<tr>
<td>FAA SA02734LA</td>
<td>AIRFILM CAMERA SYSTEMS, INC</td>
<td>FAA SA02734LA - 30.09.2020</td>
<td>Installation of Airfilm Wing Strut Utility Mount</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Airfilm Camera Systems documentation listed on STC, No. FA002734LA, dated 30 September 2020.</td>
<td>This STC is valid for aircraft types and models approved by EASA listed on the FAA Approved Model List SAA02734LA dated 30 September 2020 or later approved revisions.</td>
<td>30/03/2021</td>
<td>Active</td>
<td>10075981</td>
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<td>STC Holder</td>
<td>STC Holders</td>
<td>Rev. Level</td>
<td>Foreign/NAAC</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis for the original product as amended by the STC</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC 12A157CE</td>
<td>MYSTERE-FALCON 20 - F5</td>
<td>Electronic Flight Instrumentation System</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>Limited to MSN 392</td>
<td>31/03/2021</td>
<td>Active</td>
<td>10076109</td>
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<tr>
<td>AVIONIK STRAUBING ENTWICKLINGS GmbH</td>
<td>PA-46R-350T</td>
<td>PA-46-600TP</td>
<td>PA-46-350P</td>
<td>PA-46-20TP</td>
<td>Installation of dual Radar Altimeter Antennas Sensor Systems S67-2002</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any in...</td>
<td>Installed in accordance with &quot;instructions for installation&quot; ref:ASR-2020-064-11. ICA-09 Amendment 0 dated 22/02/2021 for later revisions of the above listed document(s) approved...</td>
<td>06/04/2021</td>
<td>Active</td>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA SR11328DS</td>
<td>AS 350 B3</td>
<td>Installation of RT-7000 RMR Tactical Radio</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>This STC is limited to Airbus Helicopters AS 350 B3 Post AMS 07-454 aircraft. 350A82-1675-01 is limited to receive-only.350A82-1675-02 enables transmit function.Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and any other previously installed change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the aeromechanics of...</td>
<td>This STC is limited to Airbus Helicopters AS 350 B3 Post AMS 07-454 aircraft. 350A82-1675-01 is limited to receive-only.350A82-1675-02 enables transmit function.Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the aeromechanics of...</td>
<td>23/06/2021</td>
<td>Active</td>
<td>10076124</td>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA SR11328DS</td>
<td>AS 350 B3</td>
<td>Installation of an Avalex DVR AVR-8345 digital video recorder</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any in...</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>13/09/2021</td>
<td>Active</td>
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<tr>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>REV. 1</td>
<td>737-8200</td>
<td>Installation of Fixed ELT structure provision and Fixed ELT External Antenna This STC consists in the installation of structural provisions for an ELT installation and installation of an ELT external antenna on Saab's SAAB 377 Connaught Fuselage, aft of Frame 787, between Stringers 1 and 2R in accordance with EL...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any in...</td>
<td>The STC EASA 10076336 (FAA SR11330DS) is limited to Airbus Helicopters AS 350 B3 Post AMS 07-454 aircraft. 350A82-1675-01 is limited to receive-only.350A82-1675-02 enables transmit function.Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the aeromechanics of...</td>
<td>The STC EASA 10076336 (FAA SR11330DS) is limited to Airbus Helicopters AS 350 B3 Post AMS 07-454 aircraft. 350A82-1675-01 is limited to receive-only.350A82-1675-02 enables transmit function.Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the aeromechanics of...</td>
<td>14/06/2021</td>
<td>Active</td>
<td>10076133</td>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA SR11328DS</td>
<td>AS 350 B3</td>
<td>Installation of a Jupiter Audio System</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>The Certification Basis for the original product as amended by the STC</td>
<td>13/06/2021</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA SR0496R6C-D</td>
<td>AS 350 B2, AS 350 B3</td>
<td>Installation of Trakteer Corporation A800 or A800MKI Searchlight</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SR0496R6C-D dated 01 August 2020. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged for this certificate/ approval without any modifications.</td>
<td>- Drawing List 350A80-1788 Rev. 1 dated 06 December 2013. For configuration 350A80-1638-04 Rotorcraft Flight Manual Supplement 350FMS-551 Rev. A dated 01 August 2020.</td>
<td>This STC is applicable to all serial numbers except the A800 MKI searchlight installation is limited to AS 350 B3 modularity. Installation of infrared light filters in the A800 or A800MKI searchlight is not authorized. The installation of A800 searchlight removable provisions (SR0A73-1570) under this STC.</td>
<td>07/04/2021</td>
<td>Active</td>
<td>10078134</td>
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<tr>
<td>KLM ENGINEERING &amp; MAINTENANCE</td>
<td>FAA STC</td>
<td>737-800</td>
<td>Cabin reconfiguration of KLM Boeing 737-800 aircrafts</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged for this certificate/ approval without any modifications.</td>
<td>- KLM Engineering &amp; Maintenance Master Data List MDL-013931 Rev 09 KLM Engineering &amp; Maintenance Aircraft Maintenance Manual AMM737-01393 1-00 original issue KLM Engineering &amp; Maintenance Aircraft Maintenance Manual AMM737-01393 1-00 original issue.</td>
<td>Concurrent installation of EASA STC 10078565 (Installation of LED Mood Lighting System) is required. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and this STC is applicable.</td>
<td>09/04/2021</td>
<td>Active</td>
<td>10076150</td>
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<tr>
<td>AEROLEDS, LLC</td>
<td>FAA STC SR02074SE</td>
<td>SEE FAA AML</td>
<td>Installation and maintenance of the Sunspot 36 and 46 series lights</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SR02074SE. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Instructions for Continued Airworthiness (ICA) as listed on FAA STC SR02074SE.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and this STC is applicable.</td>
<td>09/04/2021</td>
<td>Active</td>
<td>10076166</td>
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<td>PATS AIRCRAFT L.L.C.</td>
<td>FAA STC ST01358NY</td>
<td>737-800</td>
<td>Installation of a Lithium-ion main ship battery in accordance with FAA STC SA4303CH</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Configuration 1-4 through 1-H are limited to one year from date of issue. Incorporation of PATS, Inc. Service Bulletin Number BB22-28-08A Revision A, dated June 5, 2002, removes this limitation on effect/aircraft. Configuration 1-4 through 1-H are limited to the Aircraft Serial Number 32438, 32460...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/04/2021</td>
<td>Active</td>
<td>10076217</td>
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<tr>
<td>BLACKHAWK MODIFICATIONS Inc</td>
<td>FAA STC SA4303CH</td>
<td>B300C B300 300LW 300</td>
<td>Installation of a Lithium-ion main ship battery in accordance with FAA STC SA4303CH</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): FAA Special Condition 23-288-SCT. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Blackhawk Modifications, Inc. Master Drawing List 601-002-ML, revisionB, dated March 15, 2020. Blackhawk Modifications, Inc. Flight Manual Supplement, Document No.601-023- AFMS, Revision IR, FAA approved May 8, 2020.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/04/2021</td>
<td>Active</td>
<td>10078224</td>
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List of EASA Supplemental Type Certificates

- **AEROTEC & CONCEPT**
  - **A330-323/-341/-342/-343**
  - **A330-301/-302/-303/-321/-322**
  - **A330-201/-202/-203/-223/-243**
  - **FALCON 10**
  - Installation of a SATCOM system and antenna and integration of a WiFi system.
  - Certification basis: AE19151MOD rev. 0 dated 12th April 2021.
  - Prior to installation of this change/repair it must be determined that the STC installation in combination with any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.

- **ELBE FUSSZEUGWERK E GmbH**
  - **A330-323/-341/-342/-343**
  - **A330-301/-302/-303/-321/-322**
  - Installation of an Additional Stowage in Cargo Area of A330P2F converted aircraft.
  - Certification basis: OEO-015-001 Rev. A ENGINEERING ORDER - A330P2F FOR Cabin Stowage Cabinet(s) U.
  - Prior to installation of this change/repair it must be determined that the STC installation in combination with any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.

- **AEROTEC & CONCEPT**
  - **A330-323/-341/-342/-343**
  - **A330-301/-302/-303/-321/-322**
  - **A330-201/-202/-203/-223/-243**
  - **FALCON 10**
  - Installation of an Additional Stowage in Cargo Area of A330P2F converted aircraft.
  - Certification basis: OEO-015-001 Rev. A ENGINEERING ORDER - A330P2F FOR Cabin Stowage Cabinet(s) U.
  - Prior to installation of this change/repair it must be determined that the STC installation in combination with any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.

- **AEROLITE AG**
  - **PC-24**
  - Installation of a SATCOM system and antenna and integration of a WiFi system.
  - Certification basis: AE19151MOD rev. 0 dated 12th April 2021.
  - Prior to installation of this change/repair it must be determined that the STC installation in combination with any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.

- **SABENA TECHNICS BGC**
  - **A330-323/-341/-341**
  - **A330-321/-320/-323/-341/-342**
  - **A330-243/-245/-301/-302/-303**
  - **A330-201/-202/-203/-223/-225**
  - **FALCON 10**
  - Installation of a SATCOM system and antenna and integration of a WiFi system.
  - Certification basis: AE19151MOD rev. 0 dated 12th April 2021.
  - Prior to installation of this change/repair it must be determined that the STC installation in combination with any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.
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<th>STC Holder</th>
<th>B300 Internal Cabin Modification for ISR</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>AVIONTEC LTD.</td>
<td>B300 Internal Cabin Modification for ISR</td>
<td>A340-600 Cargo Carriage in Passenger interior reconfiguration on Boeing 737-800 series airplanes</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any limitation.</td>
<td>Master Data List GVIH-11008-01-MDL-01, Issue 14 dtd. 16-Dec-2022</td>
<td>As per Deviation D-130/0073586 EASA has established an operating limitation. The allowance to transport cargo in the passenger compartment is limited to 2000 FH following installation of the change or repair until July 31, 2022, whichever occurs first – see also Amended Limitation/Section of Document GVIH-1...</td>
<td>19/04/2021</td>
<td>Active</td>
<td>10076250</td>
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<tr>
<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td></td>
<td>A340-642, A340-643</td>
<td>The Type and OSD Certification Bases (CB) for the original product are amended by the following additional or alternative airworthiness requirements: Deviation(s): D-130/0073586/001 (B/E Aerospace Inc. - FSI) Master Drawing B/E AEROSPACE, Inc. - FSI</td>
<td>Aerotec Concept Master Data List, document AE19105MDL, Rev 1</td>
<td>In conjunction with STC 10076255 (B300 Internal Cabin Modification for ISR) and the following prerequisites: - Textron Aviation B300 aircraft with Proline Fusion avionics (B300 ER Super King Air) equipped with Dual Alt Strakes (Beechcraft Super King Air B300, F...</td>
<td>22/02/2024</td>
<td>Active</td>
<td>10076254</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>B300</td>
<td>B300 External Installations for ISR missions. Rev. 1 extends the approval to Textron Aviation B300 aircraft with PT6-67A engine and with Proline Fusion avionics (B300 ER Super King Air), and equipped with Garmin TAS, ASP-2110 and Garmin SGD-350 VHF.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any limitation.</td>
<td>MOD AE19105: External Modification for Continued Airworthiness and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/04/2021</td>
<td>Active</td>
<td>10076255</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>B300</td>
<td>B300 Internal Cabin Modification for ISR missions. Rev. 1 extends the approval to Textron Aviation B300 aircraft with PT6-67A engine and with Proline Fusion avionics (B300 ER Super King Air), and equipped with Garmin TAS, ASP-2110 and Garmin SGD-350 VHF.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Deviation(s): D-130/0073586/001 (B/E Aerospace Inc. - FSI) Master Drawing List (MDL00764501 Rev. 0)</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/04/2021</td>
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<td>STC Holder</td>
<td>Associated Technical The Certification Basis (CB) for the associated certified noise and/or emissions levels of the product</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
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### List of EASA Supplemental Type Certificates

<table>
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<tr>
<th>STC Holder</th>
<th>STC#</th>
<th>Foreign/NAA STC#</th>
<th>Reference</th>
<th>Description</th>
<th>Certification basis</th>
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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tr>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>ST03594NYH</td>
<td>FAA STC SA01869WI</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of tail tie down assembly</td>
<td>1. FAA Approved Master Drawing List as listed on AML SA01869WI, amended 23 July 2019, or later FAA Approved revision 2. Instructions for Continued Airworthiness (ICA) as listed on AML SA01869WI, amended 23 July 2019, or later FAA Approved revision.</td>
<td>This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List SA01869WI dated 25 July 2019. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/04/2021</td>
<td>Active</td>
<td>10076200</td>
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<tr>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>ST03594NYH</td>
<td>MYSTERE FALCON 900 (F900E)</td>
<td>EASA Validation of STC</td>
<td>The Certification Basis (CB) for the original product remains applicable to the certified system, approval, except where amended by or...</td>
<td>AFM Supplement FWK- D0099-150-1 Rev. IRAMS Supplement FWK- D0073-005 Rev. Eror later revisions of the above listed document(s) approved/ accepted in accordance with the Technical Implementation Procedures of ELU USA Bilateral Agreement.</td>
<td>Applicable to Falcon 900 and 900EX with EASA STC 10054966 (FAA STCST03371NY) and 10053829 (FAA STC ST03090NY) installed. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/04/2021</td>
<td>Active</td>
<td>10076207</td>
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<tr>
<td>ARMSTRONG AEROSPACE (Inc.)</td>
<td>ST03017NYD</td>
<td>FALCON 900 (F900E)</td>
<td>Certification Basis</td>
<td>The Certification Basis for the original product remains applicable to the...</td>
<td>Armstrong Aerospace Master Data List DC104-9100-00 Revision dated 22 May 2019Instruction for Continued Airworthiness and Airplane Flight Manual Supplement as listed on the Master Data List above mention later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the...</td>
<td>This STC is a prerequisite installation to ensure TPEx��明Operational approval to use T.PEDs must be obtained separately from this STC, The Luftansa BoardConnect interface with the following hardware required existing systems/components, which must be installed. The belowlisted must...</td>
<td>19/05/2021</td>
<td>Active</td>
<td>10076375</td>
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<tr>
<td>STC Holder</td>
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<tr>
<td>JET AVIATION AG</td>
<td>Installation of an adapter plate between the A/C fuselage skin and the VHF antenna, located at the forward fuselage, upper lobe between STA 900s STA 920. Including new inspection intervals Rev.1: Typo correction / editorial correction.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval without any im...</td>
<td>Engineering Order VHF Adapter Plate Modification JBLM - M-03606EMD-001Instructiona for Continued Airmaintenance VHF Adapter Plate Modification JBLM - M-03606CA-001 Iss.8Master Data List JBLM-AM-03605MDL - 001 Iss.Aor later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airmaintenance of the product.</td>
<td>20/05/2021</td>
<td>Active</td>
<td>10076305</td>
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<tr>
<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td>Installation of provisions for the EV23200 system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval without any im...</td>
<td>Declaration of compliance Document No. PP-112-DC-04 issue 4Issued 3rdMay 2021Master data list No. PP-112 MDL issue 5 dated 3rd May 2021Rorbroadcast Flight Manual Supplement No. PP-112-AFM issue 1 dated 22ndDecember 2020or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This approval is a design provision only. The activation offline EV23200and the side displays will be addressed in a separate approval. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or...</td>
<td>05/05/2021</td>
<td>Active</td>
<td>10076306</td>
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<tr>
<td>AERODATA AG</td>
<td>Installation of Automatic Flight Inspection System AD-AFIS-112</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval without any im...</td>
<td>Master Document List, Document Number MDL-A/14-1500 issue 4 dated 17-Mar-2014</td>
<td>This Installation is limited to B300, Serial Number FL-555. Operational Limitations according to AFM-176, Rev. 0 or later must be observed. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and...</td>
<td>06/05/2021</td>
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<td>10076400</td>
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<tr>
<td>TEKTRON AVIATION inc.</td>
<td>Installation of the Rockwell Collins DS-3000 system with new Primary Flight Displays, Multi-Function Display, Display Control Panels, AudioPanels and Engine Instrumentation. The installation also includesanIntegrated File Server System capable MY Satellite Graphical Weather, Turbulence Weather Radar...</td>
<td>Certification basis as defined on the original FAA STC ref SA00748DE. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval without any impact on the noise database.</td>
<td>Installation in accordance with-Hawker Beechcraft ServicesMaster Data List (MDL)014-3110- L01-00,Revision 4.0, dated 09/17/2012. Operates in accordance with -Hawker Beechcraft Services Aircraft Flight Manual Supplement (AFMS)ref014-3110- A01-50, revision 3.0, dated 09/17/2012</td>
<td>Installation of Rockwell Collins APS-65 Autopilot system must have been previously accomplished. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>05/05/2021</td>
<td>Active</td>
<td>10076410</td>
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<tr>
<td>W&amp;S CONSULTANTS, INC. (swi TELAIR INTERNATIONAL)</td>
<td>Installation of Main Deck Cargo Loading System (MDCLS)</td>
<td>The Certification Basis for the original product as amended (the following additional or alternative airmaintenance requirements the following paragraph(s) at a later amendment:CS25.302C:26.303, 26.305(a) (b),26.305(b)(b), 25.1529, 25.1541(a)(3)CS25.302C:26.307(a), 25...</td>
<td>FAA approved Telair International Index List IL904-100 revision B, dated24 February 2021, and FAA approved Airplane Flight Manual Supplement AFM904-100 revision Original, dated 18 March 2021 or later FAA approved/revision(s) FAA Approved Telair/Instructions for Continued Airmaintenance (ICA)Report No. İC...</td>
<td>The following STC are prerequisites for this installation 1) Precision Conversions, LLC Airbase A321 STC 20716-18E, 1) Telair International Cargo Floor STC STO7241A Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherprev...</td>
<td>20/05/2021</td>
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<td>10076407</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10076408</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0435AT-D</td>
<td>GV-SP GVIX</td>
<td>Installation of Iridium SATCOM System (FANS-1/A+ Over Iridium)</td>
<td>1. Gulfstream Aerospace Corporation Index List CEA1 2038308, Revision 0, dated 30 January 2018, or later FAA approved revision 2. Aircraft Flight Manual Supplement, CEA1 2038305, Revision 0, dated 30 January 2018, or later FAA approved revision 3. Aircraft Flight Manual Supplement, CEA1 2038311, Revision 0. FAA STC 10076409 FANS-1/A+ must be installed prior to or concurrently with this STC and Gulfstream G-IV-X and GV-SP FANS Over Iridium (Iridium data safety services) have not been approved to operate simultaneously with an operating Iridium SATCOM System. For aircraft equipped with an Iridium SATCOM that is interfaced...</td>
<td>08/05/2021</td>
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<td>10076410</td>
<td>ANDRA INTERNATIONAL LLC</td>
<td>FAA ST0692LA</td>
<td>737-800</td>
<td>Installation of cargo loading system</td>
<td>Anza International LLC Controlled Data List DL8360, Revision E, dated20 November 2018Informat...</td>
<td>06/05/2021</td>
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<td>10076411</td>
<td>ALPINE AEROTECH LP</td>
<td>TCCA BH21-11 ISSUE 1</td>
<td>429</td>
<td>Floor Protection Kit Installation.</td>
<td>- EASA STC 10076409 (FAA STC ST0699LA - B737-800 Passenger to Freighter Conversion) must be concurrently installed. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any/other previously installed change and/ or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>06/05/2021</td>
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<td>EASA STC No.</td>
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<td>10076414</td>
<td>STC Holder</td>
<td>REV 1</td>
<td>ATR 72-212, ATR 72-212 A, ATR 72-202, ATR 72-211 ATR 72-102, ATR 72-201 ATR 42-500, ATR 72-101 ATR 42-330, ATR 42-400 ATR 42-200, ATR 42-300</td>
<td>Installation of a Honeywell Mark VII Enhanced Proximity Warning System (EPGWS) Rev. 1. Addition of a MMEL Supplement</td>
<td>Certification basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-MMEL, issue 2: 050, 060, 100, 110, 120, 125, 130, 135, 140, 145, 150. The requirements for environmental protection and the associated certification</td>
<td>Master Data Unit PN: 0120-01-A-2103-R0 dated 07 October 2019 and FAA STC Supplement 0120-01-A-2301-R0 dated 09 November 2019.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/05/2021</td>
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<tr>
<td>10076419</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERS SERVICES INC. WALTER STARZACHER Ges</td>
<td></td>
<td>HELI AUSTRIA</td>
<td>Installation of a GTX345 Transponder</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-27.610 Amdt 4: CS-AS-CN Issue 2 Subpart A and D. The requirements for environmental protection and the associated certification</td>
<td>The change is defined with revision 1 of AMES Engineering Order Document EO_M15400114 8_RSP_23042020 The installation has to be accomplished in accordance with revision 18 of AMES Instructions for Continuous Airworthiness Document CA_M15400114 8_RSP_23042020 The aircraft shall be operated in accordance with the associated certification(s).</td>
<td>The allowance to transport cargo in the passenger compartment is limited to 2000 kg following installation of the change to 31 December 2021 whichever occurs first. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/05/2021</td>
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<td>10076440</td>
<td>AVS PMA, LLC</td>
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<td>AVS PMA, LLC</td>
<td>Approval of FAA PMA Parts for use on Bae 125 and Hawker Series Aircraft</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any implementation.</td>
<td>Approval of FAA PMA Parts under FAA RISP-9491 dated 06 November 2019.</td>
<td>In accordance with FAA-EASA TIP rev 6, Article 7.1.3, the following statement should be written in the remarks block of the FAA Form 8130-3: Produced by the holder of the EASA STC dated 1507/9440. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/05/2021</td>
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<td>10076472</td>
<td>BELL TEXTRON, Inc.</td>
<td>FAA STC SR112403</td>
<td>FAA STC SR112403</td>
<td>Installation of Accessories Package</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any implementation.</td>
<td>1. Master Drawing List, Report No. AA-18543, No Revision, dated 19 June 2019. 2. Master STC Document No. AA-1853, No Revision, dated 19 June 2019.</td>
<td>This STC is applicable to all serial numbers for the Bell Textron Canada Limited 505. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the...</td>
<td>11/05/2021</td>
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<tr>
<td>10076473</td>
<td>HELI AUSTRIA GmbH</td>
<td>AS 355 F2 AS 355 F1</td>
<td>Installation of a GXT345 Transponder</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The requirements for environmental protection and the associated certification</td>
<td>Master Document No. MDL DA-U-050-001, Revision 2, dated 10 May 2021.</td>
<td>Type of Operation: VFR only. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/05/2021</td>
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<td>EASA STC no.</td>
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<td>Rev.</td>
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<td>10076474</td>
<td>HELI AUSTRIA GmbH</td>
<td></td>
<td>AS 355 F2, AS 355 F1</td>
<td>Models 10076475</td>
<td>Installation of a G7650 COMMNAVGPS as NAV2/COMM.</td>
<td>Master Document List MDL HAU-D-051-001, Revision 2, dated 10.05.2007(Flight Manual Supplement HAU-D-051-001, Revision 2, dated 10.05.2007)</td>
<td>Type of Operation: VFR only. Prior to installation of this change/repair it must be determined that the relationships between this change/repair and any other previously installed change(s) or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>11/05/2021</td>
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<tr>
<td>10076475</td>
<td>BOEING IRELAND LIMITED</td>
<td></td>
<td>737-800</td>
<td>Models 10076479</td>
<td>Avionic provisions for a “Poland Head of State” configuration.</td>
<td>Certification Programme BIL-00189-CP rev 0 (AirCraft Flight Manual Supplement BIL-00189-AFM81 rev Instructions for Continued Airworthiness BIL-00189-ICA01 rev Limitations report BIL-00189-DR01 Rev 0) later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>11/05/2021</td>
<td>Active</td>
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<td>10076479</td>
<td>HELI AUSTRIA GmbH</td>
<td></td>
<td>AS 332 C, AS 332 L</td>
<td>Models 10076487</td>
<td>HAU-D-030 - Remote pump power supply</td>
<td>Master Document List, Doc. No. HAU-D-030-001, Rev. 0 Installation Manual, Doc. No. HAU-D-030-004, Rev. 0 Flight Manual Supplement, Doc. No. HAU-D-030-005, Rev. 0 Instructions for Continuing Airworthiness, Doc. No. HAU-D-030-006, Rev. 0 Supplemental Master Minimum Equipment List, Doc. No. HAU-D-030-511, R.</td>
<td>The following are prerequisites to the installation of this STC: 1. the helicopter must not be equipped with the “6th fuel tank” 2. the helicopter must be equipped with a cargo hook and provisions for the 4.5 ton cargo hook 3. the helicopter must embody installation of HAU-D-026, EASA MIC Approval...</td>
<td>12/05/2021</td>
<td>Active</td>
<td>10076478</td>
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<td>10076493</td>
<td>PARC AVIATION ENGINEERING SERVICESLIMITED</td>
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<td>BAE 146 SERIES 300 SEE EASA APPROVED MODEL LIST</td>
<td>Models 10076495</td>
<td>AML-25001 Installation of emergency equipment.</td>
<td>- Master Document List AML-25001-MLD Rev 02: Instructions for Continued Airworthiness AML-25001-ICA01 Rev.0 Operational Suitability Data - Cabin Crew Data Supplement AML-25001-CCD01 Rev 00: Operational Suitability Data - Boeing 737 Max Minimum Equipment List at Supplement AML-25001-MML-01 Rev 00...</td>
<td>As a result of this Change it is not permitted to install additional equipment or cabin baggage in the same compartment as the new installed portable ELT or AED. It is also not permitted to install any other pressurized equipment in the same stowage compartment. Prior to installation of this change...</td>
<td>17/05/2021</td>
<td>Active</td>
<td>10076493</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10076505</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>ST065050C-D</td>
<td>FALCON 900EX</td>
<td>Installation of Mid Cabin Cabinet Structural Installation</td>
<td>The Certification Basis for the original product remains applicable thh certificate/ approval, except where amended by revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>1. Data List Report No. F900-0106, Revision J, dated 16 April 2008, actual. FAA approved revision.</td>
<td>The design of the Luminator does not correspond to it as a module designation. This is only a commercial designation for airplane with a major modification number (M083) is embodied. The 900EX designation for Falcon 900EX does not correspond to a module designation. This is only a commercial...</td>
<td>19/05/2021</td>
<td>Active</td>
<td>10076505</td>
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<tr>
<td>10076500</td>
<td>AIRBUS DEFENCE AND SPACE S.A.</td>
<td>A330-243</td>
<td>A330-200 Emergency MEDEVAC SOLUTION</td>
<td>Installation of a Luminator search light LS-400, LS-400M, or LS-400T and to the system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.</td>
<td>The design Change Definition of this modification DCC05905E is of CS-25, paragraph 3.8.9 (a) of FAA STC. The system is certified to meet the requirements for environmental protection and the associated certified noise and/or emissions levels of the product.</td>
<td>STC EA94/S.I.0134 is pre-requisite for this installation. The EASA requirements and Falcon 900EX does not correspond to a module designation. This is only a commercial...</td>
<td>18/05/2021</td>
<td>Active</td>
<td>10076500</td>
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<td>10076501</td>
<td>LUMINATOR HOLDING L.P. d/b/a Luminator Aerospace</td>
<td>FAA STC SR09919AC</td>
<td>SEE FAA AML</td>
<td>Installation of a Luminator search light LS-400, LS-400M, or LS-400T and to the system.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Installation Instructions and Instructions for Continued Airworthiness. The attached Approved Model List Initial Release dated 06/06/2021</td>
<td>Prior to installation of this change/repair it must be determined that the airworthiness conditions for Falcon 900EX will be not affected and that the modification number (M083) is embodied. The 900EX designation for Falcon 900EX does not correspond to a module designation. This is only a commercial...</td>
<td>23/06/2021</td>
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<td>10076506</td>
<td>SABENA TECHNICS BGC</td>
<td>737-900/900ER/900/900F</td>
<td>737-600/700/800/800BC</td>
<td>Mounting device and power supply connection for portable EFB with PIVOT cases in cockpit of BOEING 737</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Master Data List, Ref. 10003-01-A-MDL-R01 dated 17 May 2021 - Airplane Flight Manual Supplement, Ref. 10003-01-A-AMFS-F01-R01 dated 12 May 2021 - Master Drawing List for Airworthiness Certification Basis for this modification - 12 May 2021, Ref. 10003-01-A-JC-F01-R00 installation dated 12 May 2021.</td>
<td>Prior to installation of this change/repair it must be determined that the airworthiness conditions for Falcon 900EX will be not affected and that the modification number (M083) is embodied. The 900EX designation for Falcon 900EX does not correspond...</td>
<td>19/05/2021</td>
<td>Active</td>
<td>10076506</td>
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<td>10076509</td>
<td>PILATUS BUSINESS AIRCRAFT, Ltd.</td>
<td>FAA STC SA00011CE</td>
<td>PC-12/47E</td>
<td>Airseat/Airseat + system installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by revision. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain applicable to this certificate/ approval.</td>
<td>Master Drawing List for Airworthiness Certification Basis for this modification - 12 May 2021, Ref. 10003-01-A-JC-F01-R00 installation dated 12 May 2021.</td>
<td>Prior to installation of this change/repair it must be determined that the airworthiness conditions for Falcon 900EX will be not affected and that the modification number (M083) is embodied. The 900EX designation for Falcon 900EX does not correspond...</td>
<td>19/05/2021</td>
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<td>10076509</td>
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<td>STC Holder</td>
<td>FAA STC/EASA STC/CAAC/UK CAA STC</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>WORTHINGTON AVIATION, LLC</td>
<td>FAA SA04467CH</td>
<td>Installation of cabin utility seats</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Worthington Aviation Inc. Drawing List 22-0026-DL, Revision 0, dated 05 June 2020 Airplane Flight Manual Supplement, 22-0026-IFMS, Revision 0, dated 24 November 2020 Instructions for Continued Airworthiness, Manual 22-0026-AM, Revision 0, dated 09 June 2020</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/05/2021</td>
<td>Active</td>
<td>10770153</td>
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<tr>
<td>ARCHEION HOLDINGS, L.L.C.</td>
<td>REV. 1 FAA STC ST514MAAT</td>
<td>Installation of Avionica satLINK MAX with optional Personality Module (PM) in the Cabin Compartment overhead, and optional arRDC MAX QAR in the EE Bay to provide 2 Iridium SATCOM ATS Voice channels. 1 Iridium SATCOM ATS DataLink FANS 1/A+ Data channel for Flight Deck use (CPDLC at RCP240 &amp; ACAS-I at RS...</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25, art 26.25:21 (a)(1), (a)(3), (a)(5), 25.303, 25.305(a) (b), 25.561(a)(6), 25.575(a)(3), 25.581, 25.601, 25.603, 25.610, 25.615.</td>
<td>1. FAA approved Airplane Flight Manual Supplement for Archeon Holdings LLC, 130716-AMFS, Revision A, dated 16 November 2018, or later FAA approved revisions</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/07/2021</td>
<td>Active</td>
<td>1077616</td>
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<tr>
<td>PILATUS BUSINESS AIRCRAFT, Ltd.</td>
<td>FAA STC SA1036DE</td>
<td>Avance L3 system installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA1036DE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Master Document List for Avance L3 in a PC-1247E, 2E-1803-01-0000, Revision B, dated 13 July 2020, or later FAA approved revisions Flight Manual Supplement, for the Installation of the Avance L3 System/for the PC-1247E. Document No. 2E-1803-65-0000, Rev - , dated 16 July 2020, or later FAA approved rev...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/05/2021</td>
<td>Active</td>
<td>1077627</td>
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<tr>
<td>JET AVIATION AG</td>
<td>PC-1247E FAA STC SA01036DE</td>
<td>VVIP Cabin Interior Completion</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS2 5-25 Amdt. 21 with revisions/exceptions as noted: 25.23, 25.25, 25.27, 25.29, 25.31, 25.301, 25.303, 25.305-25.305-25...</td>
<td>Aircraft Flight Manual Supplement JBSC-M-00156AMF-01, Issue 02; Instructions for Continued Airworthiness JBSC-M-00156AMC-01, Issue 03; Airworthiness Maintenance Tasks (AMT) (ALS) JBSC-M-00156AMT-01, Issue 02; Master Minimum Equipment List Supplement JBSC-M-00156MEL-01, Issue 02; OkoSD Cabin Crew Data JBSC-M-00156CC...</td>
<td>Private Operation (Not for Hire, Non-Commercial), Non-smoking aircraft. An approved exterior paint livery matching the exit configuration of the aircraft must be applied. Changes to the exterior marking must be approved. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/05/2021</td>
<td>Active</td>
<td>1077628</td>
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<tr>
<td>S-TEC CORPORATION</td>
<td>FAA SA09764DS</td>
<td>Installation of an S-TEC 01326 Digital Flight Control System (DFCS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA09764DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Master Data List ST-984-MDL-0001, Revision A, dated 30 March 2021 Airplane Flight Manual Supplement ST-984-AMFS-0001, Rev - , dated 31 March 2021 Instructions for Continued Airworthiness ST-984-ICA-0001, Rev - , dated 31 March 2021 and later revisions of the above listed document(s) approved/accepted on file...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/05/2021</td>
<td>Active</td>
<td>1077633</td>
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<td>STC Holder</td>
<td>FAA STC No.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>DAVID AULT FALCON JET CORPORATION</td>
<td>ST010405C-D</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of Auxature/Entertainment Center Structural Installation</td>
<td>1. Data List Report No. F05-01099 Revision AF dated 19 January 2011, or later FAA approved revision 2. For Auxature/Entertainment Center Structural Installation F0000EXand F0000EX Requirements use Report 38185, Revision 26, November 2007 3. Maintenance Manual Supplement, Report 37912, revision...</td>
<td>This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List ST010405C-D, Report No. 400686, Rev. No.(c) dated 19 January 2011. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any...</td>
<td>25/06/2021</td>
<td>Active</td>
<td>1007534</td>
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<tr>
<td>LEARJET Inc.</td>
<td>CL-600-2B16 (601-3R VARIANT)</td>
<td>Installation of a Honeywell TCZ-9 10 TCAAS II processor exchange Techware</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval, except where amended by additional or later amendments if indicated on FAA STC ST010428DE-D. The requirements for environmental protection and the associated certified noise and/or emissions levels of the...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. The FAA STC SA1077GL, SA8014NM, SA8014 and ST00689Ww...</td>
<td>10/06/2021</td>
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<td>1007535</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>A321-271NX-272NX</td>
<td>&quot;FLEX&quot; Pict Tablet Support Provision &amp; Power Supply Installation onibus SA</td>
<td>The Certification Basis for the original product as amended by following additional or alternative airworthiness requirements the following paragraphs(s) at a later amendment: CS26 and/or 4 for paragraphs CS 25.29CS 25.31(a) 25.32(a) CS 25.33CS 25.32(a) CS 25.34(a) CS 25...</td>
<td>Master Data List, Ref. 10000-1A-MLD-R90 dated 25 May 2021. (b) The FAA STCs upon the airworthiness of the product will introduce no adverse effect upon the airworthiness of the product. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/05/2021</td>
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<td>1007536</td>
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<td>AVIATION TECH ASSOCIATES LLC</td>
<td>EC 130 B4, EC 130 T2</td>
<td>Installation of Landing Gear Sleeve</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Drawing List ATA-130-110, Revision &quot;C&quot;, dated 5 August 2018. (c) FAA Approved 31 August 2016. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/08/2021</td>
<td>Active</td>
<td>1007537</td>
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<tr>
<td>LEARJET Inc.</td>
<td>CL-600-2B16 (601-3R VARIANT)</td>
<td>Installation of a MagnaStar C2000 Digital Airborne Telephone system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any...</td>
<td>FAA Approved Master Document List No. CA-1067-TUC, Revision(0), dated 28 October 1995 (=) FAA approved revisions of the above listed document(s) approved/accepted or replaced by FAA in accordance with the Technical Implementation/Procedures of EU/USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/05/2021</td>
<td>Active</td>
<td>1007538</td>
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<td>STC Holder</td>
<td>Status</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>Active</td>
<td>10076603</td>
<td>Installation of MT-PROPELLER on PA-23-250, PA-23-235</td>
<td>According to MT-Propeller Master Documentation List No. E-3317 Rev. 0-0</td>
<td>MT-Propeller ICA Document No. E-3316 Rev. 0-0</td>
<td>According to MT-Propeller Master Documentation List No. E-3317 Rev. 0-0</td>
<td>02/06/2021</td>
<td>Active</td>
<td>10076603</td>
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<tr>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>FAA</td>
<td>EMB-505</td>
<td>Installation of Garmin GI1000 NXI</td>
<td>- According to Garmin Master Drawing List MDL Number051-01385-06, Revision 1, dated: 10/29/2019 - Operate in accordance with Garmin Airplane Flight Manual Supplement190-02545-01, Rev. 1, dated: 10/29/2019</td>
<td>Garmin Master Drawing List MDL Number051-01385-06, Revision 1</td>
<td>Installation and use of Garmin Flight Stream 510 is not covered by this approval. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>02/06/2021</td>
<td>Active</td>
<td>10076603</td>
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<tr>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>EASA</td>
<td>CL-600-2B16</td>
<td>EA35 Equipment Installation of several air ambulance medical equipment provisions</td>
<td>- According to Aerolite Master Document List No. MDL-21-049Rev. 0-0</td>
<td>- Aerolite Master Document List No. MDL-21-049Rev. 0-0</td>
<td>Installation of the above listed equipment is permitted without any adverse effect upon the airworthiness of the product</td>
<td>02/06/2021</td>
<td>Active</td>
<td>10076604</td>
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<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAACertification basis</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
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<td>CL-600-2B16 (604 VARIANT)</td>
<td>Provisions for Antenna Installation in CL604</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any im...</td>
<td>Installed in accordance with Service Bulletin ref: 604/110-12-19 edition dated 12-03-2021, maintained in accordance with Maintenance ManualSupplement ref: 604/110-12-20 edition 3 dated 26-05-2021 or later revisions of the above listed document(s) approved/ accepted/undertaken by the EASA system.</td>
<td>09/06/2021</td>
<td>Active</td>
<td>10076614</td>
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<td>ST ENGINEERING AEROSPACE LTD.</td>
<td>REV. 2</td>
<td></td>
<td>A320-253N/-272N/-273N</td>
<td>CS-CCD complianceCargo-Lite Solution for A320-200 AircraftCabins reconfiguration to transport cargo in the passenger cabin/Rev. 1: Replacement of existing 4 portable oxygen cylinder/assembly with portable oxygen bottle and full face smoke mask &amp; removal of the 3 months limitation/Rev. 2: Complain...</td>
<td>installation of electrical, plumbing and ECS components only. The installation of the electrical, plumbing and ECS components only will introduce no adverse effect upon the airworthiness of the product.</td>
<td>In accordance with the Deviation D-130 and due to the type of portable oxygen equipment masks, this STC is limited to 2 000 FH following installation of the change or to July 31st 2022, whichever occurs first. Prior to installation of this change/repair it must be determined that the interrelationship between these changes/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/09/2021</td>
<td>Active</td>
<td>10076616</td>
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<td>AEROSERVIS, S.R.O.</td>
<td></td>
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<td>L 410 UVPE20 CARGO L 410 UVPE20 L 410 UVP-E</td>
<td>AS502 B, AS 502 B1, AS 360 B, AS 360 B, AS 360 D</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Deviation(s): CRI A-01, Issue 1, Rev. 2, dated 10-May-2021 The requirements for environmental protection and the associated/certified noise and/or emissions levels of the original product...</td>
<td>- Master Document List, Doc. AE 184 000 X, Rev. E, dated 20.05.2021 Certification Program, Doc. AE 184 016P, Rev. F, dated 21.05.2021 - Definition document, Doc. AE 184 115P, Rev. H, dated 02.06.2020 - L 410 UVP-E Flight Manual Supplement, Doc. STC AE184/DL1, August/2020, L 410 UVP-E Flight Manual...</td>
<td>This STC is not applicable to the aircraft equipped with an autopilot.</td>
<td>06/06/2021</td>
<td>Active</td>
<td>10076617</td>
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<td>EXPERT GEOPHYSICS LTD</td>
<td></td>
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<td>AS 350 B, AS 350 B1, AS 350 B2, AS 360 B, AS 360 D</td>
<td>Geophysics Equipment (Rad-alt and GPS antenna) Installation.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any im...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/06/2021</td>
<td>Active</td>
<td>10076625</td>
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<tr>
<td>LEARJET Inc.</td>
<td></td>
<td></td>
<td>CL-600-2B16 (601-3A VARIANT)</td>
<td>Installation of Entertainment Cabinet</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST00099WI. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the on...</td>
<td>Precision Pattern Drawing Number CA-1035-TUC stamped &quot;FAA approved&quot; with a date of 29 December 1993, or later.</td>
<td>08/06/2021</td>
<td>Active</td>
<td>10076627</td>
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<td>STC Holders</td>
<td>FAA STC</td>
<td>STOC9505W0</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>ST0005W0</td>
<td>Installation of a bulkhead across fuselage station 542 within hinging doorwing, a lavatory with a sinkand a toilet on the right hand side.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST0005W0. The requirements for environmental protection and associated certified noise and/or emissions levels of the product applicable to this certificate/approval without any...</td>
<td>Precision Pattern Drawing Number CA-1012-TUC stamped &quot;FAA Approved&quot; with date of 15 March 1993, or later.</td>
<td>03/04/2021</td>
<td>Active</td>
<td>10078630</td>
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<tr>
<td>AERO DYNAMIX, Inc.</td>
<td>FAA STC</td>
<td>ST0005W0</td>
<td>Installation of Night Vision Imaging System (NVIS) / Night Vision Goggles (NVG) compatible lighting system.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product applicable to this certificate/approval without any...</td>
<td>Instruction for Continued Airworthiness (IC) Doc ICA B407-01-Rev. B dated 10 January 2014 Master Drawing List MDL 2441-01 Rev. F dated 22 July 2020EASA Flight manual Supplement RFMSA B407-03-EASA-03 Rev.R dated 10 May 2022 for later revisions of the above listed document(s) approved/accepted in original...</td>
<td>08/06/2021</td>
<td>Active</td>
<td>10078630</td>
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<tr>
<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>ST00047W</td>
<td>Installation of a bulkhead across fuselage station 353 within hinging door and magazine racks.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST00047W. The requirements for environmental protection and associated certified noise and/or emissions levels of the product applicable to this certificate/approval without any...</td>
<td>Precision Pattern Drawing Number CA-1021-TUC stamped &quot;FAA Approved&quot; with date of 15 March 1993 or later later.</td>
<td>08/06/2021</td>
<td>Active</td>
<td>10078630</td>
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<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>ST0058AT</td>
<td>Installation Corporate Interior Cabinets</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST0058AT. The requirements for environmental protection and associated certified noise and/or emissions levels of the product applicable to this certificate/approval without any...</td>
<td>Learjet Top Drawing &quot;Arrangement Interior Cabinets&quot;, Deg No. TM313173, Revision B, dated 5 August 1994, or later FAA approved revision.</td>
<td>08/06/2021</td>
<td>Active</td>
<td>10078631</td>
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<tr>
<td>BOEING IRELAND LIMITED</td>
<td>FAA STC</td>
<td>ST0058AT</td>
<td>B737-8 Zero Pax to 189P</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any...</td>
<td>Change / Repair Approval, B737-8 Zero Pax to 189P &quot;Reconfiguration&quot; BI-00130 APP. Revision 00, dated 2nd June 2021.</td>
<td>09/04/2021</td>
<td>Approved</td>
<td>10078635</td>
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<tr>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>FAA STC</td>
<td>ST0058AT</td>
<td>WLAN Activation This Modification activates the Wireless Local Area Network (WLAN)Function of a preinstalled WLAN system/router interfaced with WAPs into the cabin emitting in 2.4GHz and/or 5GHz Frequency Band and a cockpit/WLAN ON/OFF switch.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment. For the 737-700/737-800, 737-900, 737-900ER, 737-8 the following paragraphs of CS-25 Amendment 12 are applicable...</td>
<td>Configuration 1 The Change is defined with Revision B of Master Data List document MDL- B7372-1-4430. The aircraft will be operated in accordance with Revision A of Aerial Flight Manual Supplement document AFMS- B7372-1-4430 Configuration n 2. The Change is defined with Revision A of Master Data List document...</td>
<td>31/05/2023</td>
<td>Approved</td>
<td>10078636</td>
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<td>STC Holder</td>
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<tr>
<td>LEARJET Inc.</td>
<td>ST00150W</td>
<td>Installation of bulkhead. The Certification Basis for the original product remains applicable to the certificated aircraft, approved, except where amended by additional or later amendments if indicated on FAA STC ST00150W. The requirements for environmental protection and the associated certified noise and/or emissions levels of this system.</td>
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<tr>
<td>LEARJET Inc.</td>
<td>ST00110W</td>
<td>Installation of galley assembly. The Certification Basis for the original product remains applicable to the certificated aircraft, approved, except where amended by additional or later amendments if indicated on FAA STC ST00110W. The requirements for environmental protection and the associated certified noise and/or emissions levels of this system.</td>
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<tr>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>P2006T</td>
<td>Additional gear alert system installation to inform the pilot about landing gear deployment. The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certification approval without any impact.</td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC</td>
<td>A330-302</td>
<td>Recomfiguration of the passenger cabin for transporting cargo on Iberia A330 aircraft Rev. 1. MEL supplement updated. Calendar limitation updated as follows: Aircraft modified under this STC Project can be operated in the cargo-in-cabin configuration for a maximum of 2000 flight hours or until 01-July-2024.</td>
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<tr>
<td>SABENA TECHNICS BOC</td>
<td>ATR 72-212</td>
<td>Mounting device and power supply connection for IPMOT portable EFBSolutions in cockpit of ATR42/ ATR72 Rev. 1 adds missing requirements in the Certification Basis (missing paragraphs) and updates the MSL issue number. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment number:</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>EASA STC No.</th>
</tr>
</thead>
<tbody>
<tr>
<td>CL-600-2818 (601-3A VARIANT)</td>
<td>Installation of bulkhead. The Certification Basis for the original product remains applicable to the certificated aircraft, approved, except where amended by additional or later amendments if indicated on FAA STC ST00150W. The requirements for environmental protection and the associated certified noise and/or emissions levels of this system.</td>
<td>Precision Pattern Drawing Number CA-1036-TUC stamped &quot;FAA Approved&quot; with the date of 29 September 1993, or later.</td>
<td>This approval is for installation of structural modifications only. The installation of electrical, plumbing and ECS components must be separately approved. Prior to installation of this change/repair it must be determined if the change/repair will introduce adverse effect upon the airworthiness of the product.</td>
<td>09/06/2021</td>
<td>Active</td>
<td>1007632</td>
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<tr>
<td>CL-600-2A12 (601 VARIANT)</td>
<td>Installation of galley assembly. The Certification Basis for the original product remains applicable to the certificated aircraft, approved, except where amended by additional or later amendments if indicated on FAA STC ST00110W. The requirements for environmental protection and the associated certified noise and/or emissions levels of this system.</td>
<td>Precision Pattern Drawing Number CA-1037-TUC stamped &quot;FAA Approved&quot; with the date of 29 September 1993, or later.</td>
<td>This approval is for installation of structural modifications only. The installation of electrical, plumbing and ECS components must be separately approved. Prior to installation of this change/repair it must be determined if the change/repair will introduce adverse effect upon the airworthiness of the product.</td>
<td>10/06/2021</td>
<td>Active</td>
<td>1007639</td>
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<tr>
<td>A330-202</td>
<td>Additional gear alert system installation to inform the pilot about landing gear deployment. The Certification Basis (CB) for the original product remains applicable to this certification approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certification approval without any impact.</td>
<td>Master Document List, S4A Solutions for Aviation document No.P2006T12-19-02, Edition., dated 11 June 2021. Flight Manual Supplement, S4A Solutions for Aviation document No.P2006T12-12-28, Edition 01, dated 21 August 2020. Maintenance Manual Supplement, S4A Solutions for Aviation document No.P2006T12-12...</td>
<td>Prior to installation of this change/repair it must be determined if the change/repair will introduce adverse effect upon the airworthiness of the product.</td>
<td>14/06/2021</td>
<td>Active</td>
<td>1007666</td>
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<tr>
<td>A330-302</td>
<td>Recomfiguration of the passenger cabin for transporting cargo on Iberia A330 aircraft Rev. 1. MEL supplement updated. Calendar limitation updated as follows: Aircraft modified under this STC Project can be operated in the cargo-in-cabin configuration for a maximum of 2000 flight hours or until 01-July-2024.</td>
<td>Certification Summary, N0689-D90, Revision A, dated 26 July 2021. Flight Manual Supplement, N0689-D85, Revision C, dated 27 May 2021. ICA, N0689-D70, Revision B, dated 17 May 2021. Maintenance Manual Supplement, N0689-D86, Revision F, dated 22 July 2020.</td>
<td>Aircraft modified under this STC Project can be operated in the cargo-in-cabin configuration for a maximum of 2000 flight hours or until 01-July-2024. Whatever occurs first. Prior to installation of this change/repair it must be determined if the change/repair will introduce adverse effect upon the airworthiness of the product.</td>
<td>30/07/2021</td>
<td>Active</td>
<td>1007697</td>
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<tr>
<td>ATR 72-212, ATR 72-211</td>
<td>Mounting device and power supply connection for IPMOT portable EFBSolutions in cockpit of ATR42/ ATR72 Rev. 1 adds missing requirements in the Certification Basis (missing paragraphs) and updates the MSL issue number.</td>
<td>- Master Data List, Ref. 10002-01-A-MDL-R01, dated 18 June 2021.</td>
<td>Limitations as per §3 of Aircraft Flight Manual Supplement referenced above. Prior to installation of this change/repair it must be determined if the change/repair will introduce adverse effect upon the...</td>
<td>22/06/2021</td>
<td>Active</td>
<td>1007678</td>
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<tr>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>BOEING IRELAND LIMITED</td>
<td>Installation of Business Class Mini-Suites</td>
<td>Certification Programme ref. BIL-00103-CP Revision 04Declaration of Compliance ref. BIL-00103-</td>
<td>APP Revision 00 dated 18.06.2021 Aircraft Flight Manual Supplement ref. BIL-00103-</td>
<td>At time of approval of this change to type certificate, the approval described parts of</td>
<td>21/06/2021</td>
<td>Active</td>
<td>10076728</td>
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<tr>
<td>ZIM AIRCRAFT SEATING GMBH</td>
<td>STC for EC1 A330 Air SeatbackNew plastic materials for backrest shells and kick panels. The design change covers the parts to ensure continued airstream of the delivered seats.</td>
<td>Certification for embodiment of a certification by EASA STC 10062195 or STC 10063950 or STC 10059695 or STC 10066025. Prior to installation of this change/repair it must be determined that the configuration described between this change/repair and any other previously installed modification is safe.</td>
<td>Certification basis ref. BIL-00103-CP Revision 04Declaration of Compliance ref. BIL-00103-</td>
<td>APP Revision 00 dated 18.06.2021 Aircraft Flight Manual Supplement ref. BIL-00103-</td>
<td>23/06/2021</td>
<td>Active</td>
<td>10076742</td>
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<tr>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>Installation of ELT’s on B737/Installation of Automatic Fixed ELT within the passenger cabin.</td>
<td>Flight Manual Supplement FMS/1329-001 Issue 2</td>
<td>Master Document List MDL/1329-001 Issue 10SD Flight Crew Data Supplement FCDG/1329-001 Issue 3or later revisions of the above listed document(s).</td>
<td>Operations with this STC installed are limited to 2 000 FH (following issuance of the STC) or to July 31st 2022, whichever occurs first. Cabin crew operating aeroplanes modified per this STC are considered to be a “variant” to the EASA STC for EC1 A330 Air Caraibes New.</td>
<td>22/06/2021</td>
<td>Active</td>
<td>10076736</td>
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<tr>
<td>APTOZ EHF</td>
<td>Conversion of main cabin to cargo area/This change converts the main passenger cabin to cargo area and after removal of cabin arrangements. REV. 1 of this STC approval extends the calendar limitation to July 31st 2022. The referenced documents have been updated accordingly.</td>
<td>Aspire Flight Manual Supplement document number: AFM251171-1 at issueG dated October 4th, 2021</td>
<td>Master Data List/1329-001 Issue K dated October 4th, 2021</td>
<td>In addition/Rev. 00 dated October 4th, 2021Declarations of Compliance...</td>
<td>26/10/2021</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
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<td>Associated Technical Documentation</td>
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<td>10076478</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST03117AT-D</td>
<td>GIV-X</td>
<td>Installation of Iridium Satellite Telephone System</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>GC412003K, Rev A, dated April 28, 2021</td>
<td>EASA Validation Summary Compliance Report – GIV-X Iridium Satellite Telephone System/GC412003K</td>
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<tr>
<td>10076750</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>FAA ST02047SE</td>
<td>777-300ER</td>
<td>Installation of a new seats and In Flight Entertainment system.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>N9339-O10, Rev D, dated 22 January 2010 or later</td>
<td>FAA Aviation Administration (FAA) approved revisions.</td>
</tr>
<tr>
<td>10076757</td>
<td>PILATUS BUSINESS AIRCRAFT, Ltd.</td>
<td>REV. 1 FAA SAU0069AT</td>
<td>PC-12, PC-1245, PC-1247</td>
<td>Installation of an Equipment Console, interfacer: 1 connects the FAA STC number.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Drawing List No. BR-96002, Revision C, dated 24 July 2018</td>
<td>FAA approved revisions.</td>
</tr>
<tr>
<td>10076760</td>
<td>AVIATION SPECIALTIES UNLIMITED, INC.</td>
<td>REV. 1 FAA SB02512SE</td>
<td>EC 225 LP AS 332 L</td>
<td>Initial Revision: EASA validation of the Night Vision Imaging System(s) (NVIS) FAA Supplemental Type Certificate (STC) ref. SR09921AC for the EC225 LP helicopter model Revision 1: Addition of AS 332 L model</td>
<td>The Type and OIS Certification Bases (CB) for the original product remain unchanged except where amended by additional or later amendments if indicated on FAA STC SR09921AC. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Initial Revision: Project Specific Certification Plan (PSCP) ref. ASU-332-22-27-16 Revision C, dated 04/05/2021., Rotorcraft Flight Manual (RFM) Supplement ref. FMS-332A/S 4 Revision B, dated 17/05/2021 and approved by the FAA on 21/06/2021, Instructions for Continued Airworthiness (ICA) document ref. I-1, 24/03/2017 Revision1: Design change limited to AS 332 L helicopters with asc06...</td>
<td>Initial Revision: Design change limited to EC225 LP helicopters with autonavigation described in the ICA-332004-4 Appendices Revision B dated 16/06/2021, supplementing the ICA document ref. ICA-332004-4 Revision A dated 24/03/2017 Revision1: Design change limited to AS 332 L helicopters with asc06...</td>
</tr>
<tr>
<td>10076761</td>
<td>LIBERTY PARTNERS, INC.</td>
<td>FAA STC SR09921AC SEE FAA APPROVED MODEL LIST</td>
<td>Demonstration of aircraft critical equipment and antenna system/maintenance of radio frequency interference (RFI) from intentional on-air transmissions from certain Portable Electronic Devices (PEDs)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SR09921AC. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the or...</td>
<td>1. Master Data: list LP9293- DL001, revision IR, dated 15 January 2021, 2. One of the following Rotorcraft Flight Manual Supplements (RFMS) inequed as defined in the master data list: LP9293-FMK01 Revision IR, dated 1 February 2021, LP9293- FMK02 Revision IR, dated 1 February 2021, LP9293- FMK03 Rev...</td>
<td>This STC does not constitute installation approval of a wireless radio frequency system using wireless data technologies. It only provides that data that may partially support its installation or its activation. This STC is valid for aircraft types and models approved by EASA and digitised on the FAA Approved M...</td>
<td>06/06/2021</td>
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**List of EASA Supplemental Type Certificates**

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<thead>
<tr>
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<tr>
<td>LIBERTY PARTNERS, INC.</td>
<td>737-800</td>
<td>Cargo on passenger cabin floor</td>
<td>The Certification Basis for the original product remains applicable to certificated aircraft approval, except where amended by additional or later amendments if indicated on FAA STC. The requirements for environmental protection and the associated certified noise emission levels of the or...</td>
<td>1. Master Drawing List in Appendix 2. Airplane Flight Manual Supplement-&lt;br&gt; STC- EM-10001 Revision B. - LP23-02-003 Revision B -&gt; LP23-FM013 Revision B.</td>
<td>None</td>
<td>29/06/2021</td>
<td>Active</td>
<td>10076770</td>
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<tr>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>BD-700-1A10</td>
<td>CREW REST SEAT LIMITATIONS CHANGE BD700The STC consists in the crew rest seat.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Paragraph(s) at a later amendment: 26.1, 26.1, 26.301, 26.8, 26.301, 26.9, 26.301. The requirements for environmental protection and the limited to 2 000 FH after installation or its activation.</td>
<td>GEXS5-RPT02 Master drawing list in Appendix 1</td>
<td>None</td>
<td>29/06/2021</td>
<td>Active</td>
<td>10076771</td>
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<tr>
<td>UAB &quot;FL TECHNICS&quot;</td>
<td>737-800</td>
<td>Cargo on passenger cabin floor</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Paragraph(s) at a later amendment: 26.1, 26.1, 26.301, 26.8. The requirement for environmental protection and the associated certified noise emission levels of the or...</td>
<td>MDL-527-01 Issue K – Master Data List/PA-527-03 Issue H – Aircraft Flight Manual Supplement/PA-527-04 Issue G – Flight Attendant Manual Supplement/PA-527-01 Issue G – Change/Repair Bulletin later revisions of the above list document(s) approved by EASA.</td>
<td>None</td>
<td>07/12/2021</td>
<td>Active</td>
<td>10076772</td>
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<tr>
<td>SIRIUM AEROTECH S.L.</td>
<td>GIV-S</td>
<td>Activation ADS-B OUT Functions for Gulfstream GIV/GV</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Paragraph(s) at a later amendment: EASA CS-ACNS Subpart D, Section 4. The requirements for environmental protection and the associated certified noise emission levels of the or...</td>
<td>SA20-63S-AMMS Rev01, SA20-63S-MDL Rev01, SA20-63S-MDL Rev01, SA20-63S-MDL Rev01, SA20-63S-MDL Rev01</td>
<td>None</td>
<td>30/06/2021</td>
<td>Active</td>
<td>10076774</td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>A320-214</td>
<td>Installation of seats, monuments and SPS</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - Paragraph(s) at a later amendment: JAR 25 chapter 13: 25.861(a)(1)(i)(I), 25.861(a)(1)(i)(II), 25.869(a)(1)(I). The requirements for environmental protection and the associated certified noise emission levels of the or...</td>
<td>Master Drawing List (MDL) in Appendix 4. - Installation of seats, monuments and SPS.</td>
<td>- installations of 618NT001-1 through 65 require prior or concurrent installation of the STC 10008645 (FAA STC 5078498) and STC 10008645 (FAA STC 5078498).</td>
<td>01/07/2021</td>
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<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>SIRIM AEROTECH S.L.</td>
<td>Installation of Idradim DLU system and interface with existing Datalink equipment for FANS 1/A.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impacts.</td>
<td>SA21-26S-MDL Rev 01, SA21-26S-AMS Rev 01, SA21-26S-MMELs Rev 01(&gt;0) or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>-</td>
<td>01/07/2021</td>
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<td>DOA AESGLOBAL LIMITED</td>
<td>ACTIVATION OF BLUEBOX WIFI SYSTEM Rev 1: applicability listextend for A318-111/-113/-115/-131/-132/-133/-151N/-153N/-171N.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements - The following paragraph(s) at a later amendment: CS-25 Amend. 23, Sections 25.1, 25.561, 25.601, 25.605, 25.606, 25.611, 25.787, 25.789, 25.831, 25.853, 25.869...</td>
<td>For models A320-211/-212/-214/-215/-216/-231/-232/-233/-235N/-2372N/-272N/-273N</td>
<td>Prior to installation of this change aircraft configuration must comply with concurrent requirements defined in section 5.1 of Master Data List AES-320-577MDL. Prior to installation of this change/repair it must be determined if/whether this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/03/2024</td>
<td>Active</td>
<td>10078001</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>Interior Outfitting: The change consists of a complete cabin interior outfitting of the A320-251N with 34 passengers VIP and 5 cabin crew seats for private use only.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) of CS-25 at Amendment 21 are applicable to this change: CS25.1, CS25.20, CS25.21, CS25.35, CS25.35, CS25.27, CS25.29, CS25.301, CS25.303, CS25.305...</td>
<td></td>
<td>Private use only. Prior to installation of this change/repair it must be determined if/whether this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2023</td>
<td>Active</td>
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<td>ETHAD AIRWAYS ENGINEERING L.L.C.</td>
<td>CABIN RECONFIGURATION FROM 249 TO 250P: Installation of additional economy class seats in zones between doors 1 and 3 to increase passenger seating capacity to have 235 economy passengers. PAIX configuration (post-mod configuration: &quot;Aircraft A&quot;). Revision 1 of this STC introduces a design change to increase passenger...</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impacts.</td>
<td>Ethihad Airways Engineering L.L.C. Master Document List EASA-32151420-DL01/Issue 2</td>
<td>Prior to installation of this change/repair it must be determined if/whether this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/02/2022</td>
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<td>EASA STC no.</td>
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<td>Certification Basis</td>
<td>Certification Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>1007815</td>
<td>AIRBUS S.A.S.</td>
<td>Airbus Interiors Services MOD A90104</td>
<td>Airbus Certification</td>
<td>Airbus Certification</td>
<td>This STC covers 3 possible configurations - TITAN/TCS 195</td>
<td>04/07/2021</td>
<td>Active</td>
<td>1007815</td>
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<td>1007816</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FALCON 900EX (F900EX)</td>
<td>Maintenance Manual</td>
<td>Maintenance Manual</td>
<td>This approval is limited to only the installation made in DassaultAviation Model 900EX Serial No. 223 Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse...</td>
<td>02/07/2021</td>
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<td>1007826</td>
<td>SET FLUGTECHNIK GmbH</td>
<td>KODIAK 100</td>
<td>Installation documentation</td>
<td>Installation documentation</td>
<td>Minimum flight crew: when mission system is in operation one pilot/crew in the left cockpit seat and one operator required in the cabin. Maximum occupant seating limit: 3 (ref. to AFMS-KOD100-9501-01 + Other previously installed change and/or repair will introduce no adverse...</td>
<td>05/07/2021</td>
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<td>1007883</td>
<td>PATRIA AVIATION OY</td>
<td>429</td>
<td>Maintenance Manual</td>
<td>Maintenance Manual</td>
<td>Flight into known icing conditions is prohibited. Minimum flight crew: when mission system is in operation one pilot/crew in the left cockpit seat and one operator required in the cabin. Maximum occupant seating limit: 3 (ref. to AFMS-KOD100-9501-01 + Other previously installed change and/or repair will introduce no adverse...</td>
<td>06/07/2021</td>
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<td>1007888</td>
<td>HONEYWELL INTERNATIONAL Inc.</td>
<td>REV. 2</td>
<td>Installation of the Honeywell RDR-7000 Volumetric Radar Weather System</td>
<td>Installation of the Honeywell RDR-7000 Volumetric Radar Weather System</td>
<td>This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List dated 12 April 2022. Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse...</td>
<td>17/06/2022</td>
<td>Active</td>
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<td>STC no.</td>
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<td>Associated Technical Documentation</td>
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<td>10076898</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES N.C.</td>
<td>777-300ER</td>
<td></td>
<td></td>
<td>Cabin reconfiguration of British Airways 777-300ER</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification note(s) and/or emissions levels of the product are unchanged.</td>
<td>Master Drawing List, Northwest Aerospace Technologies, Inc. document N00345-010, Revision 0, dated 28 June 2021 Instructions for Continued Airworthiness (ICA) Document No. 5753-0008-00, Rev. 0, dated 04 November 2020 for later revisions of the above listed document(s) approved/accepted...</td>
<td>09/07/2021</td>
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<tr>
<td>10076896</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST01943WI</td>
<td>SEE FAA AML</td>
<td></td>
<td>Modification of Rockwell Collins, Inc. Multi-Mode Receiver (MMR) Global Landing System (GLS-2100) Software installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification note(s) and/or emissions levels of the original product are unchanged...</td>
<td>(1) Approved Model List, RCA-0663, Revision (1), issued 17 April 2005 (2) Instructions for Continued Airworthiness, RCA-0656, Revision 0, dated 2 March 2020 or later FAA approved/accepted revision for items (1) and (2)</td>
<td>16/08/2021</td>
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<td>10076900</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>BD-700-1A10</td>
<td></td>
<td></td>
<td>Rockwell Collins Venue RestRoft/Existing CES system components performing audio/video signal selection, distribution, backlit displays, Galley Touchscreen display, DVO/players, audio speakers and amplifiers will be removed and replaced with units from Venue system performing same function.</td>
<td>Certification Report BB07-QCM23-014-M Initial issue-ICA BB07-QCM2014-014-075 19/07/2021</td>
<td>09/07/2021</td>
<td>Active</td>
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<td>10076901</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>GV-SP</td>
<td>GV</td>
<td>Installation of a KA Broadband High Speed Data System, installed onGulfstream model GV &amp; GV-SP (G550) Rev. 1: correct typo</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification note(s) and/or emissions levels of the product are unchanged.</td>
<td>FAA STC ST04220AT-DDGL100000031 Index List Rev ADDL0000001758 Config List Rev BGCD000019368 Rev –, EASA Validated Configuration List -ST04220AT -DCE5193030325 Cert Plan Rev ACE5193030341 Summary Compliance Report Rev – CES19303041 AFMS Rev –CES19303041141 ICA Rev Bor later revisions of the above listed doc...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previsouly installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10076908</td>
<td>TURK HAVA YOLLARI TEKNIK A.E., TURKISH TECHNOLOGY INC.)</td>
<td>737-800</td>
<td></td>
<td></td>
<td>737 – Cabin Refurbishment Phase 2This project includes the installation of a windshield, curtain rail, literature pockets, baby bassinet and a megaphone.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification note(s) and/or emissions levels of the product are unchanged.</td>
<td>Certification Programme CS25-CEP-02, dated 18th June 2021 section 6 take 20 or later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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**List of EASA Supplemental Type Certificates**

**EASA STC no.**

<table>
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<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>AVOIDINE CORPORATION</td>
<td>FAA STC SA0041BO</td>
<td>SEE FAA AMIL</td>
<td>Installation of a SkyTrax 100B/200 ADS-B Receiver 700-0190-100</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA0041BO. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and are applicable to this certificate/approval without any modification.</td>
<td>Master Document List (MDL) AVMILB-002, Revision 04, dated 31 May 2021. In accordance with 14CFR 121.305(a)(1).</td>
<td>This approval is for the installation of the SkyTrax 100B/200 receiver and associated antenna only. Separate approval of a proven compatible display system is necessary in order to use the received information. There is no AFMS associated with this STC. This approval does not cover the installation of the associated certified noise and/or emissions levels of the original product.</td>
<td>12/07/2021</td>
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<td>10075911</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 2</td>
<td>FAA STC ST04338AT-D</td>
<td>SEE FAA AMIL</td>
<td>Installation of an Air Ionizer System on Gulfstream Models: G280, G-IV, G-IV-X, GV, GV-SP and GVI</td>
<td>Gulfstream Aerospace FAA Approved Model List (AML). Installation of the Air Ionizer System on Gulfstream Models: G280, G-IV, G-IV-X, GV, GV-SP and GVI. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and are applicable to this certificate/approval without any modification.</td>
<td>This STC applies to aircraft with smoke detection systems that utilize photovoltaic method of smoke detection. Additional compliance findings for §§ 25.858(a) are required for aircraft configured with smoke detection systems utilizing any other method of detection except photovoltaic method. Prior to installation this STC must be determined that there is no adverse effect upon the airworthiness of the aircraft.</td>
<td>15/07/2021</td>
<td>Active</td>
<td>10075913</td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>FAA STC ST02865OSE</td>
<td>A320</td>
<td>ECOS Enlarged Shelf Bin Installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC ST02865OSE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and are applicable to this certificate/approval without any modification.</td>
<td>Northwest Aerospace Technologies, Inc. Master Drawing List N0667-D10, Revision D, dated 26 June 2021, or later. Federal Aviation Administration (FAA) approved revision. Instructions for Continued Airworthiness (ICN) N0667-D70, Revision A, dated 27 April 2021, or later FAA approved revision.</td>
<td>Prior to installation of this STC it must be determined that there is no adverse effect upon the airworthiness of the aircraft.</td>
<td>13/07/2021</td>
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<td>10075917</td>
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<tr>
<td>CESSNA ZURICH CITATION SERVICE CENTERMBH</td>
<td>510</td>
<td>Cargo Trans Platform / passenger cabin</td>
<td>Installed in accordance with installation instruction nft: ICS-224 Rev-A, maintained in accordance with Instruction for Continuous Airworthiness nft: ICA-224 Rev. IR. Operated in accordance with Aircraft Flight Manual (supplement) nft: FMS-224 Rev. IR or later revisions of the above listed document (s).</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. Prior to installation of this STC it must be determined that there is no adverse effect upon the airworthiness of the aircraft.</td>
<td>13/07/2021</td>
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<td>KNSI AVIATION ENGINEERING LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A330-223</td>
<td>Cabin reconfiguration from a current LOPA of 24 B/C seats and 258 E/C seats to a post modification configuration of 361 E/C seats. Revision 1 includes the UK STC reference.</td>
<td>KNSI Master Document List No: 19665-MDL-001, Revision 02KNSIAircraft Maintenance Manual Supplement No. 19K865-AMM0031 Revision 02KNSI Cabin Crew Data No. 19K865-CCD-001 Revision 02KNSI Instrument Configuration Data No. 19K865-ICD-001 Revision 02KNSI</td>
<td>Prior to installation of this STC it must be determined that there is no adverse effect upon the airworthiness of the aircraft.</td>
<td>29/07/2021</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<td>LUFTHANSA</td>
<td>A330-302</td>
<td>REV. 2</td>
<td>A330-201/-202/-203/-223/-243</td>
<td>Provisional Cargo Compartment REV. 1: Calendar limitation extended to 31 July 2022 REV. 2: Model applicability extended to A330-201/-202/-203/-223/-243 with a second configuration</td>
<td>The Certification Basis for the original product as amendable: featuring additional or alternative airworthiness requirements: Deviation(s): - DEV D-0076546: &quot;Transportation of cargo in passenger compartments&quot;. The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>- Supplemental Type Definition Document, Ref. FAA STC00-2021M, issue 02 (Rev. 01 dated 17 Dec 2021) For A330-300: - Airplane Flight Manual Supplement, Ref. A05-2103-AFM Issue 08 Rev. 00 dated 24 Nov 2021: Instructions for Continued Airworthiness, Ref. A05-2103-ICA Issue 03 (Rev. 05 dated 26 Nov 2021: Cabin C...</td>
<td>The allowance to transport cargo in passenger compartments is limited to 2000 FH following installation of the change oro 31 July 2022, whichever occurs first. Aeroplanes modified per this STC are considered to be a &quot;variant&quot; to the A330-200-300 passenger configuration. Cabin crew operating aeroplane...</td>
<td>17/12/2021</td>
<td>Active</td>
<td>10079622</td>
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<tr>
<td>DELTA ENGINEERING</td>
<td>A330-941</td>
<td>REV. 1</td>
<td>FAA STC ST04076NY</td>
<td>Installation of a goGo 2Ku radome/antenna (with Demod option/Rev. 1) to change the Maximum Take-off Weight (MTOW) for the A330-941 weight variants.</td>
<td>The Certification Basis for the original product as amendable: featuring additional or alternative airworthiness requirements: Equivalent Safety Finding(s): - B-4301-005 &quot;Vibration and buffeting: Compliance criteria for large external antenna installation&quot; Issue 2 The requirements for environmental protection and/or emissions...</td>
<td>Delta Engineering Master Data List 0154-10622-2465 Revision M, dated 23 August 2021&lt;(...)&gt; Instructions for Continued Airworthiness 0754-10623-2465 Revision H, dated 5 March 2021&lt;(...)&gt; Airplane Flight Manual Supplement 0654-10624-2465 Revision C&lt;(...)&gt; dated 28 April 2021&lt;(...)&gt; or later revisions of the...</td>
<td>The activation of the goGo 2Ku antenna is not covered by this approval. This STC cannot be installed on Airbus A330-900 weight variants 920, 921 and 922. Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will not impact the airworthiness of the aircraft.</td>
<td>26/02/2024</td>
<td>Active</td>
<td>10079632</td>
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<tr>
<td>ALITALIA</td>
<td>A330-301</td>
<td>REV. 2</td>
<td>777-200, 777-300</td>
<td>Reconfiguration for transport of cargo in the cabin. This project removes cabin seats to allow transportation of cargo pallets on passenger floor REV. 2 of this STC approval extends the calendar limitation to July 31st, 2022. The referenced documents have been updated accordingly.</td>
<td>The Certification Basis for the original product as amendable: featuring additional or alternative airworthiness requirements: Deviation(s): - OR D-01 &quot;Transportation of cargo in passenger compartments&quot;. The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Airplane Flight Manual Supplement, document number: AFMS-DOA-2021-0001 at issue 2 dated December 14, 2021 Master Data List, document number: NDL-DOA-2021-0001 at issue 01 dated December 14, 2021 Instruction for Continued Airworthiness, document number: 2021-0001-CD-ICA-01 at issue 01 dated December 14...</td>
<td>Operations with this STC installed are limited to 2,000 FH following issuance of the STC or to July 31, 2022, whichever occurs first. Cabin crew operating aeroplanes modified per this STC are considered to be a &quot;variant&quot; to the B777-200/-200LR-300/-300ER passenger configurations. Cabin crew operating aeroplane...</td>
<td>17/12/2021</td>
<td>Active</td>
<td>10079644</td>
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<tr>
<td>AEROSPACE DESIGN &amp; COMPLIANCE LLC</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>REV. 2</td>
<td>FAA STC ST0406NY</td>
<td>Installation of Viasat VMT-1500 VR-12C Ku Band SATCOM System</td>
<td>The Certification Basis for the original product as amendable: featuring additional or alternative airworthiness requirements: Deviation(s): - Master Data List, Ref. CL-600-010, Revision 1, dated March 2021 Instructions for Continued Airworthiness (ICA), CL-600-ICA-0081, Revision 1, dated March 2021 Airplane Flight Manual Supplement (AFMS), CL-600-AMFS-0081, Revision 1, dated March 2021 for later revisions of the above listed document(s) a...</td>
<td>This STC requires installation of FAA STC ST0406NY for Mounted Radome. Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness...</td>
<td>19/07/2021</td>
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<td>STC Holder</td>
<td>A340 - Cabin Reconfiguration for the transportation of cargo in passenger compartments</td>
<td>- Master Data LL, Ref: 49696-03-A-MLD-R03 dated 05 Oct 2021 - Airplane Flight Manual Supplement</td>
<td>Limitations as per §3 of Instructions for Continued Airworthiness document and §3 of Airplane Flight Manual Supplement - Refer to Table 1</td>
<td>11/10/2021</td>
<td>Active</td>
<td>10776053</td>
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<tr>
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<td>Description</td>
<td>A340 - Cabin Reconfiguration for the transportation of cargo in passenger compartments</td>
<td>- Master Data LL, Ref: 49696-03-A-AMFS-R01 dated 05 Oct 2021 - Airplane Flight Manual Supplement</td>
<td>Limitations as per §3 of Instructions for Continued Airworthiness document and §3 of Airplane Flight Manual Supplement - Refer to Table 1</td>
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<td>Description</td>
<td>A340 - Cabin Reconfiguration for the transportation of cargo in passenger compartments</td>
<td>- Master Data LL, Ref: 49696-03-A-OISDCD-R01 dated 12 Jul 2021 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>- For the design covered by this approval, the following limitation applies: Oxygen Discharge Provisions. Not certified for use with pressurized Oxygen. &gt;&gt;The installation of the associated placard is done under Modification MOD-S-00-4321 CHOP 1 MINOR MODIFICATION The EASA validation of FAA STC...</td>
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<td>10079052</td>
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<td>Master Document List SDL-S-00-4174 Issue BMMEL Supplement MMELS-B3036-00-125 issue DAFM Supplement AFMS-B3036-99-009 issue Finstructions for Continued Airworthiness ICA-B3036-00-032 issue for further consideration.</td>
<td>- Master Document List SDL-S-00-4174 Issue BMMEL Supplement MMELS-B3036-00-125 issue DAFM Supplement AFMS-B3036-99-009 issue Finstructions for Continued Airworthiness ICA-B3036-00-032 issue for...</td>
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<td>When the C-MUSIC pod is not installed - A placard must be installed in lieu of the above document: The C-MUSIC Control Panel must be powered on and the two FPIM annunciators must be marked NO/Green. &gt;&gt;The installation of the C-MUSIC pod is prohibited - The C-MUSIC must be powered in TTOL, in order to have correct DIPR...</td>
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<td>Master Document List SDL-S-99-4182 Issue DAFM Supplement AFMS-B3036-99-009 issue Finstructions for Continued Airworthiness ICA-B3036-99-310 issue EAirworthiness Limitations Section document ALS-B3036-99-315 issue GMMEL Supplement MMELS-B3036-99-316 issue for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>When the C-MUSIC pod is not installed - A placard must be installed in lieu of the above document: The C-MUSIC Control Panel must be powered on and the two FPIM annunciators must be marked NO/Green. &gt;&gt;The installation of the C-MUSIC pod is prohibited - The C-MUSIC must be powered in TTOL, in order to have correct DIPR...</td>
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<td>Limitations as per §3 of Instructions for Continued Airworthiness document and §3 of Airplane Flight Manual Supplement - Refer to Table 1</td>
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<td>- Master Data LL, Ref: 49696-03-A-OISDCD-R01 dated 12 Jul 2021 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>- Master Data LL, Ref: 49696-03-A-OISDCD-R01 dated 12 Jul 2021 for later revisions of the above listed document(s) approved/accepted under the...</td>
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### Notes
- **STC Holder**: The company or entity that holds the Supplemental Type Certificate.
- **Description**: Details about the modification or change being certified.
- **Certification basis**: References to specific documents and standards that support the certification.
- **Limitations/Conditions**: Details on any limitations or conditions associated with the STC.
- **Date of Issue**: The date when the STC was issued.
- **Status**: Indicates whether the STC is active or not.
- **System Record**: Unique identifier for the STC within a system or database.
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<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Foreign/NAA STC ref.</th>
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<th>Certification basis</th>
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<td>10076967</td>
<td>T.P.S. S.P.A.</td>
<td>AB139, AW139</td>
<td>New neonatal transport system for AB139 P/N TPS139-07-005; change introduces the installation of an incubator equipped with medical devices and oxygen cylinders.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any impact.</td>
<td>- Flight Manual Supplement TPS139-0700-RFM5-00- Master Document List TPS139-0700-M6-00- Master Drawing List TPS139-0700-M7-00- Instructions for Continued Airworthiness TPS139-0700-ICA-00- Installation Service Bulletin TPS139-0700-ST0-000A later revisions of the above listed document(s) approved/approval(s).</td>
<td>- The usage of the validation data of this STC is limited to the revalidation of FFS and/or FTD devices simulating the AW139 aircraft on the basis of the AW139 equipped with Avionics Phase 3.0 configuration and fixed landing gear (i.e. Leonardo Rotorcraft Flight Manual Supplement No. 9): Devices to be assessed.</td>
<td>22/07/2021</td>
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<tr>
<td>10076972</td>
<td>328 SUPPORT SERVICES GmbH</td>
<td>AW169</td>
<td>328 Support Services OSO-SIM Data for AW169 Full Flight Simulators</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - the paragraph(s) at a later amendment CS 590 initial issued 02.12.2014.</td>
<td>Master Drawing List MDL-1342002 issue B, dated 20.07.2021, Validation Data Report: Flight Test Report Leonardo AW169-SR-02B00020001 issue E, dated 28.05.2021.</td>
<td>- Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product. There are no airworthiness limitations for this certificate.</td>
<td>21/07/2021</td>
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<td>10076981</td>
<td>KILROY AVIATION LLC</td>
<td>FAA STC ST04644NY</td>
<td>Installation of Avionic Instruments' Frequency Converters in an Airbus Helicopters Model GVI (G560/G560ER)</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval, except where amended by additional or later amendments if indicated on FAA STC ST04644NY. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>1. Kilroy Aviation, LLC, Master Data List - Installation, JSB-030-001, revision E, dated 7 May 2021, or later FAA approved revision. 2. Kilroy Aviation, LLC, Instructions for Continued Airworthiness (ICA), document JSB-ICA, revision A, dated 1 May 2021, or later FAA approved revisions.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product. There are no airworthiness limitations for this certificate.</td>
<td>23/07/2021</td>
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<td>10076986</td>
<td>AIRCRAFT TECHNICAL DEVELOPMENT Inc.d/b/a AEROCON ENGINEERING COMPANY</td>
<td>FAA STC ST01143LA</td>
<td>Installation of Aercon Engineering company (AEC) Airstair</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval, except where amended by additional or later amendments if indicated on FAA STC ST01143LA.</td>
<td>AEC Master Drawing List (MDL) 1423-3700, Revision C, dated 30 January 2001, or later FAA approved revisions.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/08/2021</td>
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<tr>
<td>10076990</td>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FAA ST0904NY</td>
<td>MISTRERE FALCON 900 (MF900) FALCON 900EX (F900EX)</td>
<td>EASA Validation of STC ST0904NY for Installation of a Honeywell ACM 8000 KA Band SATCOM in Dassault Aviation Mystere-Falcon 900 and Falcon 900EX</td>
<td>- Master Data List FXWX- DD0087-005 Rev. B-ARM Supplement FXWX- DD0087-150 Rev. IR-ICA FXWX-DD0087-150 Rev. for later revisions of the above listed document(s) approved/approved addition/or addition(s)</td>
<td>This STC is applicable to DA MF-900 and 900EX with radome installed per Dassault Aviation MD 5044 and DA 900EX with radome installed per Dassault Aviation MD 6067 or Service Bulletin 5S4 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair...</td>
<td>23/07/2021</td>
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### List of EASA Supplemental Type Certificates

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<th>STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
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<th>Model</th>
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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<td>10077003</td>
<td>JANMOMS, INC.</td>
<td>FAA STD ST01894AT</td>
<td>737-700</td>
<td></td>
<td>Relocation of High Frequency Transceiver 1 and 2 (HF1 and HF2), Cockpit Voice Recorder (CVR), Flight Data Recorder (FDR), and Auxiliary PowerUnit Electronic Control Unit (APUECU), and approval of the E8 Electronics Rack.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data Unit No. F0131 Rev. C dated 27 April 1999, for SN 22000 (registration no. N742P8) only, and for all other installations Rev. F, dated 29 July 1999 or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>27/08/2021</td>
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<tr>
<td>10077005</td>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA ST093745C</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td></td>
<td>Installation of Passenger Oxygen Distribution for the Bombardier model CL-600-2B16 (604 Variant)</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>- ST093745C, Canadair CL-600-2B16, Installation of a Passenger OxygenDistribution System Certificate: Instructions for Continued Airworthiness for Passenger Oxygen system JMP10-005 dated August 20, 1997 (revisions on Apr 20/98 and Dec 18/00). Airplane Flight Manual Supplement For Canadair Challenger...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>26/07/2021</td>
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<td>10077006</td>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA ST06943CH</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td></td>
<td>Traffic Alert and Collision Avoidance System (TCAS) Modification - Installation of Lower Directional Antenna and Relocation of Lower Transponder Antenna</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Drawing List No. 694-5305-1, Rev. F, dated 02 March 1999/Installation of Lower Antenna, Maintenance Manual Supplement KCA-AMS-02(=)(=) dated 03 April 1999 or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures.</td>
<td>Coltino TCAS-94 must have been previously installed or to be installed concurrently with this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect.</td>
<td>26/07/2021</td>
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<td>10077009</td>
<td>SABENA TECHNICS BOO</td>
<td>A330-243</td>
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<td>Forward cabin modification &quot;S0005 Maritime&quot;.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Master Document List MDL S.25-4034 rev AAFM Supplement AFMS- S0005-25-009 rev Construction for Continued Airworthiness ICA- S0005-25-007 rev ANMEL Supplement MMELS- S0005-25-10 rev Cor later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures.</td>
<td>Limitations as per AFM Supplement AFMS- S0005-25-009.Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10077012</td>
<td>AERSALE, Inc.</td>
<td>FAA STC ST04184NY</td>
<td>777-200LR 777-200</td>
<td></td>
<td>Installation of ACSS DO-260B ADS-B Out Qualified Transponder Automatic Dependent Surveillance-Broadcast (ADS-B) Out System. The validated STC (ST0141MN) is fully compliant to EASA CSACS Subpart D. Section 2 (Mode C Primary Surveillance), Section 3 (Mode SEnhanced Surveillance) and Section 4 (1090ES).</td>
<td>The Certification Basis for the original product is amended by the following additional or alternative airworthiness requirements: CS.ACS/CS Subpart D, Section 2.3 &amp; 4.7(a) the requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged...</td>
<td>Master Data List, Doc Ref: B772-MLD-005 Rev 8Airplane Flight Manual Supplement, Doc Ref: B772-105-005 Rev 19Detailed Instructions for Continued Airworthiness, Doc Ref: B772-ICA-005 Rev (Cor later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures.</td>
<td>None</td>
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<td>10077017</td>
<td>ANAVI OPERATIONS LLC</td>
<td>FAA STC 60528CH</td>
<td>A321-213/233/232 A321-115/112-131/211-212</td>
<td>Installation of a broadband wireless internet system</td>
<td>The Certification Basis for the original product remains applicable to both this installation and any associated certification noise and/or emissions levels of the original product.</td>
<td>Master Data List. Document No. RW/105-0003-00, Revision 0, dated 26 May 2021. Project Specific Certification Plan (PSCP) RW/105-0002-00 Rev. C, dated 26 May 2021.</td>
<td>The EASA STC 10073636 Rev. 0 (FAA STC 6462CH - Installation of a Broadband Antenna System) Provision is applicable to this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10077018</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC 805944DE</td>
<td>S25C</td>
<td>Installation of Concorde VRLA battery</td>
<td>The Certification Basis for the original product remains applicable to both this installation and any associated certification noise and/or emissions levels of the original product.</td>
<td>1. Document No. 5-0654, Master Drawing List, Concorde 253C (C4) Battery Replacement, Revision NC dated 19 February 2019, FAA approved 12 March 2019 or later FAA approved revision 2. Instructions for Continued Airworthiness (ICA). Document No. 5-0655, Revision NC dated 10 February 2019, FAA accepted 12 March 2019.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10077019</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC 801076DE</td>
<td>SR22T</td>
<td>Installation of main battery replacement with Concorde Valve Regulated Acid battery</td>
<td>The Certification Basis for the original product remains applicable to both this installation and any associated certification noise and/or emissions levels of the original product.</td>
<td>1. Document No. 5-0712, Master Drawing List, Cirrus SR22T Battery Replacement, Revision NC dated 20 October 2020, FAA approved 2 November 2020, FAA approved revision 2. Instructions for Continued Airworthiness (ICA). Document No. 5-0714, Revision NC dated 20 October 2020, FAA accepted 3 November 2020.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>PILATUS BUSINESS AIRCRAFT, Ltd.</td>
<td>FAA STC 605634DE</td>
<td>PC-12/47E PC-12/47 PC-12/45 PC-12</td>
<td>Installation of a SPECTRE Lift Platform System and ventral camera turetprovision</td>
<td>The Certification Basis for the original product remains applicable to both this installation and any associated certification noise and/or emissions levels of the original product.</td>
<td>1. Master Drawing List, Document No. PBA-600.103-MLX, Revision J, dated 21 October 2015, FAA approved 27 October 2015. FAA approved revision 2. FAA approved revision 2. Airplane Flight Manual Supplement for the modified Model PC-12/45 and PC-12/47 aircraft. Document PBA-600.103-MLS, Revision J. 2. Master Drawing List, Document No. 5-0704, Revision NC, dated 31 October 2015, FAA accepted 12 November 2015.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10077027</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC SR01067DE</td>
<td>480B 480</td>
<td>Installation of main battery replacement with Concorde Lead-Acid battery</td>
<td>The Certification Basis for the original product remains applicable to both this installation and any associated certification noise and/or emissions levels of the original product.</td>
<td>Document No. 5-702, Master Drawing List, Enstrom 408 &amp; 408B Battery Replacement, Revision NC, dated 14 August 2020, FAA approved 5 October 2020, FAA approved 8 October 2020. FAA approved revision 2. FAA approved revision 2. Airplane Flight Manual Supplement for the modified Model PC-12/45 and PC-12/47 aircraft. Document PBA-600.103-MLS, Revision J.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10077043</td>
<td>INNOVATOR, LLC</td>
<td>Installation of a modified horizontal stabilizer cover assembly and anti-embraic provisions</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA0469CH. The requirements for environmental protection and the associated vehicles noise and/ or emissions levels of the or...</td>
<td>1. Innovative LLC Installation Data List INH-11, IOD-20008 rev. IR dated 15 April 2020, or later FAA-Approved Revision. 2. General Arrangement Drawing INH-11-5501-001 rev. IR dated 2 April 2020, or later FAA-Approved Revision 3. Instructions for Continued Airworthiness, INH-11-ICA-20704 rev. IR dated 2 April 2020.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/08/2021</td>
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<td>10077054</td>
<td>BOEING (IRELAND) LIMITED</td>
<td>Change of the Electrical Power System to Provide 60 Minutes Standby Power for the Airplane Standby Loads in Lieu of 30 Minutes and Change/Cabin's PIP Heat power source to the Standby Bus.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-25 amdt 12 for the following paragraphs: 23252733(a)(b)(3033055)(b)(307 (a23)(31)(33333374196)(a71 (a3)(359)(a)(3)(68160...</td>
<td>Change/Repair Approval document BIL-00114-APP Rev 00, dated 6 July 2021 for Instructions for Continued Airworthiness BIL-00114-CA01 rev 00, dated 20 May 2020 Aircraft Flight Manual Supplement BIL-00114-AMM1 Rev 00, dated 10 March 2021 Fight Crew Operations Manual Supplement BIL-00114- M812 Rev 00, dated 30 Apr...</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change/repair will introduce...</td>
<td>03/06/2021</td>
<td>Active</td>
<td>10077054</td>
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<td>10077075</td>
<td>MAXCRAFT AVIONICS LTD</td>
<td>DC Electrical Panel Rele...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any im...</td>
<td>Master Data List M9L-05-70-00-031 Rev. E dated Sep 21, 2020/CA-06-70-00-031 Rev. C dated 6 March 2013, For AS 355 E, F, F1 F2, FFS-05-70-00-031(E1) Rev. A dated Aug 13, 2020 For AS 365 N MFS-05-70-00-031(E2) Rev. A dated Aug 13, 2020 or later revisions of the above listed document(s) (k) approved/app...</td>
<td>Approved for Day VFR operations only. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/08/2021</td>
<td>Active</td>
<td>10077075</td>
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<td>10077081</td>
<td>IKHANA GROUP, LLC</td>
<td>Installation of High-Flow 250 Amp Starter-Generators</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval, except when amended by additional or later amendments if indicated on FAA STC SA02708A. The requirements for environmental protection and the associated vehicles noise and/ or emissions levels of the or...</td>
<td>IKhANA Aircraft Services Master Drawing ER18-30611, Revision NG. dated 11 July 2020, or later FAA approved revision FAA Approved Flight Manual Supplement, Document No. SG-62-1-260 Revision IR dated 3 August 2020, or later FAA approved revision is required for the DHC-6-200 airplane model FAA Appr...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product. This STC approval is limited to DHC-6-200 and DHC-6-300...</td>
<td>11/08/2021</td>
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<td>1007806</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td></td>
<td>F/A SR02719SE</td>
<td>AS 350 B3</td>
<td>Onboard System Cargo Swing Suspension System for AS 350 B3 fitted with Crash Resistant Fuel Tank System (CRFS) as per TCH MOD 07/20034</td>
<td>The Certification Basis (CRB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/required noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impacts.</td>
<td>Master Document MOD No. 155-216-00, Rev 2, dated 30 March 2021; Instructions For continued Airworthiness No. 123-031-00-Rev. 00, dated 4 March 2021; Rotorcraft Fight Manual Supplement No. 121-071-00; Rev. 00, dated 27 April 2021; later revisions of the above listed document(s) approved/accepted in the relevant order.</td>
<td>-</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/06/2021</td>
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<td>1007706</td>
<td>SCANDINAVIAN AVIONICS - DESIGN ApS</td>
<td></td>
<td>GULFSTREAM G150</td>
<td>Dual CVFR Installation</td>
<td>The Certification Basis for the original product as amended by the following: Special Condition (s): CRI F-01, Non-rechargeable Lithium Batteries Installation. The following paragraph(s) applies at a later amendment: following paragraphs a)CS 25 Amendment 2; CS 25.</td>
<td>Certification documents (MDL) Doc. No. 2129299 DOC, Issue 1AFMS, Doc. No. 2129299-AM, Issue 2 and Doc No. 2129299-AM 8, Issue 1 CONTINUED AIRWORTHINESS. Doc. No. 2129299-CA, Issue 1MAINTENANCE MANUAL SUPPLEMENT, Doc. No. 2129299-AM, Issue 1MAEL SUPPLEMENT, Doc. No. 2129299-ME, Issue 1TECHNICAL ORDER</td>
<td>-</td>
<td>Limited to serial number 0283/392 to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/08/2021</td>
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<td>10077099</td>
<td>AIRCRAFT COMPLETION ENGINEERING REV. 3</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td></td>
<td>ACE STC001 - Oxygen medical bottle</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): CRI F-01, Non-rechargeable Lithium Batteries Installation. The following paragraphs apply at a later amendment: following paragraphs a)CS 25 Amendment 2 and CS 25.60</td>
<td>Master Document MOD No. 1901-2-001; Issue 1 dated 24/08/2024; Fight Manual Supplement ACE STC 001_AFMS Supplement Rev 0 dated 12/08/2021</td>
<td>-</td>
<td>EASA Minor Changes 10072988, 10083368 and 10084565 are pre-requisite for this installation. This installation is limited to configurations as per Service Bulletin ACE-001-4-005 OXYGEN MEDICAL KIT REV 8 or Service Bulletin ACE-001-3-001 Alternative support for ACE medical oxygen kit EdPr to install.</td>
<td>10/06/2024</td>
<td>Active</td>
<td>10077099</td>
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<tr>
<td>10077101</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td></td>
<td>747-8</td>
<td>Installation of Self Protection System - CAMPS</td>
<td>The Certification Basis for the original product as amended by the following: Additional or Alternative Airworthiness Requirements: Special Condition (s): CRI F-01 Missile Protection System Installation. The following paragraph(s) apply at a later amendment: the following paragraphs of CS-25 at Amendment 2 apply a</td>
<td>-</td>
<td>The change is defined with Revision B of Master Data List document MDL-B7471-1-4300. The aircraft shall be operated in accordance with Revision A of Aircraft Fight Manual Supplement document APMS-B7471-1-4300. The STC grants no operational approval, the aircraft operator will need to obtain operational approval from the Agency or Authoritative responsible for the State of registry and the State in whose airspace it is to be operated. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/08/2021</td>
<td>Active</td>
<td>10077101</td>
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<tr>
<td>10077113</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td></td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Installation of Self Protection System - CAMPS</td>
<td>The Certification Basis for the original product as amended by the following: Additional or Alternative Airworthiness Requirements: Special Condition (s): D-01 Medical Evacuation Configuration; the following paragraph(s) apply at a later amendment: CS-25.1; CS-25.21; CS-25.23; CS-25.25; CS-25.301; CS-25.303; CS-25.304</td>
<td>-</td>
<td>Installed in accordance with Service Bulletin ref 60492-12-19 edition 2 dated 17/8/2021; maintained in accordance with Maintenance Manual Supplement ref 60492-12-30 edition 1 dated 27/10/2020; operated in accordance with Flight Manual Supplement ref 60492-12-28 edition 2 dated 24/02/2021 and Operating.</td>
<td>-</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/08/2021</td>
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List of EASA Supplemental Type Certificates

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<tr>
<td>LEARJET Inc.</td>
<td>REV. 4</td>
<td>FAA STD ST01970WI</td>
<td>SEE FAA ANL</td>
<td>Installation of an Aircraft Health Management System (AHMS) Revision 1 consists of addition of Challenger 604, 605, 650, 650Global, 600 Global Vision Flight Deck and Special Condition for Systems/electrical (25-794-SC) for Challenger models and activation of RS-422 on Challenger 300, 350 installs.Revision...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/ approval without any impact...</td>
<td>Refer to FAA Approved Model List later revisions of the above listed document(s) approved/ accepted/under EASA in accordance with the Technical Implementation/Procedures of EU/USA Bilateral Agreement.</td>
<td>This STC is valid for aircraft types and models approved by EASA and listed on the FAA Approved Model List ST01970WI dated 2 August 2023. This STC is applicable to serial numbers listed in MDL BAA00015001. Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>27/06/2023</td>
<td>Active</td>
<td>10077114</td>
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<tr>
<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>REV. 1</td>
<td>737-800</td>
<td>189 Passenger Configuration</td>
<td>This change reconfigures cabin interior of Boeing 737-800 aircraft astidows:-Removal of exiting economy class seats.- Installation of new Acro seats.- Re-placing of existing FUs to suit new LOPA.- Installation of new portable ELT.- Installation of emergency &amp; safety eq...</td>
<td>The Certification Basis (CB) for the original product remains applicable/this certificate/ approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/ approval without any impact...</td>
<td>Master Data List ref. GVH-10495-01-MDL-01 at issue 1 dated 20 Aug 2021/Instructions for Continued Airworthiness ref. GVH-10495-01-CA-01 at issue 1 dated 29 Apr 2021/or later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>02/06/2021</td>
<td>Active</td>
<td>10077124</td>
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<tr>
<td>ANORDA INTERNATIONAL LLC</td>
<td>FAA STC ST03757A</td>
<td>ATR 72-212 A</td>
<td>Installation of main deck cargo loading system</td>
<td>The Certification Basis for the original product as amended/this certificate/approval is amended by the following: CS 25 anndt. 0: 25.23 (a)(b),25.23(c)(a), 25.303, 25.471, 25.501, 25.607, 25.607, 25.619,25.625, 25.1529, 25.1641/CS...</td>
<td>1. FAA approved Ancra International LLC Controlled Data List DL85440/Revision C, dated 14 June 2021, or later FAA approved revision/2. Airplane Flight Manual Supplement for ATR72-212A, report number1300, Revision R, dated 25 July 2021, or later FAA approved revision 3. Instructions for Continued A...</td>
<td>ATR Major modifications 5648, 6521, and 5977 must be installed in conjunction with this Installation on model ATR 72-212A (&quot;500 version&quot;) Full Cargo defined by TCD: EASA.A.044/Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>27/06/2021</td>
<td>Active</td>
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<td>AEROMETALS INC.</td>
<td>FAA STC SR02806LA</td>
<td>EC 130 T2</td>
<td>Installation of Engine Intake Barrier Filter System/Installation of FAA STC SR02806LA</td>
<td>The Certification Basis for the original product as amended/this certificate/approval is amended by the following: CS 25 anndt. 0: 25.23 (a)(b),25.23(c)(a), 25.303, 25.471, 25.501, 25.607, 25.607, 25.619,25.625, 25.1529, 25.1641/CS...</td>
<td>Aerometals Master Drawing List, doc. no. 210-3164-1, Rev. 8, dated June22, 2017/Instructions for Continued Airworthiness/Engine Intake Barrier SystemArrows/EC130/rev. 3, doc. no. 201-3164-1, Rev. B, dated July 18,2018/ EASA Approved Rotorcraft Manual Supplement (RFMS), doc. no. 210-3164-1, RFMS-2, dated...</td>
<td>The limitations specified in the basis Rotorcraft Flight Manual are amended by the following:- Flight in falling and/or blowing snow with IFR bypass doors closed and controlled, - Maximum operating altitude in flight: 16,000 ft pressure altitude.- ordinary altitude, whichever is lower/- Engine Health...</td>
<td>27/06/2021</td>
<td>Active</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1</td>
<td>GV-SP G5</td>
<td>Installation of JetWave Ka HSD System with Mid Fin Cap Radome for GV-SPunch</td>
<td>The Certification Basis (CB) for the original product remains applicable/this certificate/approval. The requirements for environmental protection and the associated/certificated noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any impact...</td>
<td>Supplemental Type Certificates ST04035A-T-D, reissued May 26,2020/Instructions for Continued Airworthiness/Engine Intake Barrier SystemGULFSTREAM Model GV-SP/IN-01, Revision (1), dated May 02, 2019/Instructions for Continued Airworthiness/Engine Intake Barrier SystemGULFSTREAM Model GV-SP/IN-01, Revision (2), dated May 02, 2019</td>
<td>Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>01/06/2021</td>
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<td>Date of Issue</td>
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<td>SKY SERVICE F.B.O., INC.</td>
<td>LEARJET MODEL 45</td>
<td>STC0 UL-801 UNILINK Airborne Data Link</td>
<td>The Certification Basis for the original product as amended by following/additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: FAR 25.1457(a) amendment 25-02-C. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any in...</td>
<td>Installation must be in accordance with Aeronica Design Ltd. Master Drawing List MDL18060 Revision 1C, approved September 10, 2019.</td>
<td>31/08/2021</td>
<td>Active</td>
<td>10077717</td>
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<td>APTOZ EH</td>
<td>Installation of Techtest Limited Portable ELT 500-32</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>The Certification Basis for the original product as amended by following/additional or alternative airworthiness requirements. As per Approved Model List. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any in...</td>
<td>Approved Model List AML21147 rev 3 Master Data List MDL21147 rev 3 Car later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>31/08/2021</td>
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<td>10077714</td>
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<td>2 EXCEL DESIGN LIMITED</td>
<td>B200</td>
<td>Installation of Flight Inspection Antennas to King Air B200 Rev. 1 and Rev. 2 issued for editorial corrections.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any in...</td>
<td>Installed in accordance with 2Excel Aviation Limited Installation/Instructions No. 2XL178 LFT73(1) Issue 2.</td>
<td>21/09/2021</td>
<td>Active</td>
<td>10077176</td>
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<td>HONEYWELL INTERNATIONAL Inc</td>
<td>FAA ST04473CH</td>
<td>SEE FAA AML</td>
<td>Installation of a Honeywell RDR-7000 Weather Radar System on Embraer S.A. EMB-135BJ listed from FAA AML Item 2.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any in...</td>
<td>09/06/2022</td>
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<td>10077188</td>
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<td>ICELANDAIR EHF</td>
<td>757-200</td>
<td>Structural provisions of removed SATCOM structural provisions for the High Gain Antenna &amp; Low Noise Amplifier-Diplexer unit and for the equipment rack in the forward cargo hold.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any in...</td>
<td>- Certification Plan CP-757-5310-209 Rev A - Instructions for Continued Airworthiness (Limitations section) ICA-757-5910-209-07 Rev I/Ranged 11. Aug 2013</td>
<td>Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/06/2021</td>
<td>Active</td>
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<td>HONEYWELL INTERNATIONAL Inc</td>
<td>FAA ST04470CH</td>
<td>INSTALL RDW-7000 Weather Radar on MF/90 Installation of the Honeywell RDR-7000 Weather Radar System on Dassault Aviation Mystere-Falcon 900.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval without any in...</td>
<td>- ATEC-077-241-1000 STC Master Data List Rev 0.</td>
<td>Prior to installation of this change/repair it must be determined the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/06/2022</td>
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<td>10077190</td>
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<td>Foreign/NAA STC ref.</td>
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<td>Certification basis</td>
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<td>HONEYWELL INTERNATIONAL Inc.</td>
<td>REV. 2</td>
<td>FAA STC 04477CH</td>
<td>680 (SOVEREIGN)</td>
<td>Installation of Honeywell RDR-7000 Radar STC for the installation of the Honeywell RDR-7000 Weather Radar System on Teledon Aviation Cessna 680 (Soveroign). The Type and OSD Certification Bases (CB) for the original product are applicable to this certificate/approval...</td>
<td>ATEC-377-320 STC Master Data List Rev. FATEC-20513 Airplane Flight Manual Supplement rev 1ATEC-20513 Airplane Flight Manual Supplement rev 1ATEC-21109 Airplane Flight manual Supplement rev 1 for later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with...</td>
<td>This STC is applicable to the following serial numbers for those Make-Model: MSN 680-001 to 680-050. Prior to installation of this change/repair it must be determined that interference between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>25/04/2022</td>
<td>Active</td>
<td>10077110</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 3</td>
<td>FAA STC 70424/CH</td>
<td>777-300ER</td>
<td>Installation of a Gogo Cabin Distribution System with 2G/ Satellite Connectivity:Revision 1: Mention of the Alternative Method of Compliance (AMOC) to Airworthiness Directive (AD) 2014-11/03:Revision 2: Transfer of the STC from INTELSAT INFLIGHT LLC to DeltaEngineering Corporation and editorial updates...</td>
<td>1. Delta Engineering Master Data List (MDL) D12525-21620 Revision 10 dated 2023/3 Instructions for Continued Airworthiness, Electrical Wiring/Interconnect System Instructions for Continued Airworthiness, Doc. No D12454-21620, Revision V, dated 19 July 2023/3, FAA Approved Airplane Flight Man...</td>
<td>Reference the Master Data List for additional limitations and conditions as well as applicable serial numbers:Operation approval to use T.PED’s must be obtained separately from this STC. Prior to installation of this change/repair it must be determined that interference between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>27/10/2023</td>
<td>Active</td>
<td>10077119</td>
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<tr>
<td>DASSAUT FALCON JET</td>
<td></td>
<td>FAA STC 709315C-D</td>
<td>FALCON 2000, FALCON 2000EX</td>
<td>Installation of the Ordinance Signs</td>
<td>Drawing List Report No. 7F2M/0154, Revision NC, dated 24 January 2002 or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that interference between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>08/09/2021</td>
<td>Active</td>
<td>10077114</td>
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<tr>
<td>ALAMO ENGINEERING GmbH</td>
<td></td>
<td>EMB-145MP</td>
<td>EMB-145MP, EMB-145SL, EMB-145EU, EMB-145ER, EMB-145EP, EMB-145, EMB-135ER, EMB-135SL, EMB-135BJ</td>
<td>Installation in accordance with: Aor later FAA approved top level software 7H-13683-10 or later FAA approved top level software. Prior to installation of this change/repair it must be determined that interference between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>-Master Document List, MDL21A16AE-01, Rev. A. Aircraft Flight Manual Supplement, FMS21A16AE-01, Rev. A. Instructions for Continued Airworthiness, ICAI21A16AE-01, Rev. A0r later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that interference between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>06/04/2021</td>
<td>Active</td>
<td>10077110</td>
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<tr>
<td>INNOVATIVE SOLUTIONS &amp; SUPPORT, Inc.</td>
<td></td>
<td>FAA SA04090B</td>
<td>B300, B300C, B200, B200C, B200COT, B200COT</td>
<td>Installation of FAA STC SA04090B for the following configurations:STC Configuration 2: Innovative Solutions &amp; Support, Inc. (ISR++V=10) (inflight) Standby Unit (SU) with ATIS, DualEngine, installed in 2A:B200, B200C, B200COT, B200COT or 2B:B300, B200C, B200COT, B200COT, B300B300C. All original parts...</td>
<td>-1. To be installed according to STC Configuration 2/ISU Part Number 9K-88150-1, loaded with top level software 7H-13685-10 or later FAA approved top level software. In accordance with: ICAI21A16AE-01, Rev. A. Aircraft Flight Manual Supplement, FMS21A16AE-01, Rev. A. Instructions for Continued Airworthiness, Doc. No D12454-21620, Revision V, dated 19 July 2023/3, FAA Approved Airplane Flight Man...</td>
<td>Configuration 1 of FAA STC SA04090B is not included in this approval. Prior to installation of this change/repair it must be determined that interference between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>06/06/2021</td>
<td>Active</td>
<td>10077108</td>
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<tr>
<td>BROMMA AIR MAINTENANCE AB</td>
<td></td>
<td>ATEC-077-241-1200</td>
<td>ATEC-077-241-1200</td>
<td>Installation of FAA STC SA04090B for the following configurations:STC Configuration 2: Innovative Solutions &amp; Support, Inc. (ISR++V=10) (inflight) Standby Unit (SU) with ATIS, DualEngine, installed in 2A:B200, B200C, B200COT, B200COT or 2B:B300, B200C, B200COT, B200COT, B300B300C. All original parts...</td>
<td>-1. To be installed according to STC Configuration 2/ISU Part Number 9K-88150-1, loaded with top level software 7H-13685-10 or later FAA approved top level software. In accordance with: ICAI21A16AE-01, Rev. A. Aircraft Flight Manual Supplement, FMS21A16AE-01, Rev. A. Instructions for Continued Airworthiness, Doc. No D12454-21620, Revision V, dated 19 July 2023/3, FAA Approved Airplane Flight Man...</td>
<td>Configuration 1 of FAA STC SA04090B is not included in this approval. Prior to installation of this change/repair it must be determined that interference between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>06/09/2021</td>
<td>Active</td>
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<tr>
<td>LEARJET MODEL 45</td>
<td></td>
<td>Learjet 45 Medavac Rack Installation</td>
<td>Learjet 45 Medavac Rack Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval... The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval... The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any...</td>
<td>This STC is applicable to the following serial numbers for those Make-Model: MSN 680-001 to 680-050. Prior to installation of this change/repair it must be determined that interference between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>08/06/2021</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10077207</td>
<td>DASSAULT AIRCRAFT SERVICES (DASSAULT)</td>
<td></td>
<td></td>
<td>FALCON 7X</td>
<td>Installation of a GoGo AVANCE L5 system in Dassault Aviation Falcon 7X aircraft.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or alternative airworthiness requirements or associated certified noise and/or emissions levels of the product.</td>
<td>Dassault Aviation Services (1) Master Data List F7XXV-D0084-005, Rev. IR, dated 16 February 2018; (2) Instructions for Continued Airworthiness (ICA) F7XXV-D0084-155, Rev. IR, dated 16 February 2018, or later FAA accepted revision. The ICA contents must be incorporated.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/02/2021</td>
<td>Active</td>
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<tr>
<td>10077247</td>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td></td>
<td>FAA STC ST0485AT</td>
<td>A321-271NX,-272N,-272NX</td>
<td>Installation of the Avionics Support Group Constant Friction Mount(about Mount), Cradle and Integrated Power Supply (IPS) on Airbus A318, A319, A320 and A321 Series Aircraft.</td>
<td>The Certification Basis for the original product was amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25.a. (i), 25.613(c)(26) amdt. 1: 25.611, 25.1301(c)(25) amdt. 18: 25.773, 25.1353, 25.1357, 25.1360(c)(25) amdt. 19: 25.571.</td>
<td>ST04520AT FAA STC</td>
<td>The Certification Basis for the original product is unchanged and remains applicable to this certificate/ approval without any amendment.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/09/2021</td>
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<tr>
<td>10077254</td>
<td>3 POINTS AVIATION CORP.</td>
<td></td>
<td>TCIA SA12-42</td>
<td>DHC-8-315</td>
<td>Replacement of worn upper caps of cockpit seat tracks with stainless steel track.</td>
<td>The Certification Basis (CB) of the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any amendment.</td>
<td>DHC-8-314 DHC-8-311 DHC-8-301 DHC-8-202 DHC-8-201 DHC-8-106 DHC-8-103 DHC-8-102</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/09/2021</td>
<td>Active</td>
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<tr>
<td>10077275</td>
<td>BOMBARDIER Inc.</td>
<td>REV. 1</td>
<td>TCIA SA20-92</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Emergency Equipment Modification</td>
<td>The Certification Basis (CB) of the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any amendment.</td>
<td>CL-600-2B16 (601-3R VARIANT) CL-600-2B16 (601-3A VARIANT) CL-600-2A12 (601 VARIANT) CL-600-1A11 (600)</td>
<td>According to B/E Aerospace Limited Modification Data Summary N100000, Rev N100005 was maintained by TCIA. The TCIA approved Bombardier Inc. Instructions for Continued Airworthiness document BAE3058100-ICA-XXX (Where XXX is a unique identifier for each configuration defined in Modification Data Summary N100000) is approved.</td>
<td>See those of TCIA STC SA20-92.The FAA approved configurations are identified in the Modification Data Summary N100000, Rev N10005 or later TCIA accepted revision.Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/09/2021</td>
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<tr>
<td>10077279</td>
<td>B/E AEROSPACE, INC.</td>
<td>FAA STC ST04520AT</td>
<td>SEE FAA AML</td>
<td>Installation of Rockwell Collins GLU-2100 Multi Mode Receivers (MMR)</td>
<td>The Certification Basis for the original product was amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25.a. (a), 25.303(a)(b), 25.303, 25.305(a), 25.307(a), 25.561(a), 25.601, 25.603(a)(c), 25.611, 25.850(a), 25...</td>
<td>FAA Approved Model List (AML) 2-157186-002, Revision IR, dated 21 June 2021 or later FAA accepted revisions.Instructions for Continued Airworthiness (ICA) as listed in the FAA Approved Model List (AML) document and the Aircraft Flight Manual Supplement (AFMS) as listed in the FAA Approved Model List (AML) document.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/09/2021</td>
<td>Active</td>
<td>10077279</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issuance</td>
<td>Status</td>
<td>System Record</td>
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<td>10077284</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SR11334DS</td>
<td>AS 350 B3</td>
<td>Installation of a Flightcell DZMx Satellite/Cellular/Tracking System. The Certification Basis (CB) for the original product remains applicable to this certificate/ approval except where amended by additional oralter amendments if indicated on FAA STC SR11334DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of TH...</td>
<td>Master Drawing UH 350A42-1917, rev. O, dated 17 January 2023; Rotorcraft Flight Manual Supplement 350FMS-565, rev A, dated 9 August 2020; Instructions for Continued Airworthiness 350FMS-006, rev. O, dated 23 March 2022; and/or later revisions of the above listed document(s) approved/accepted by EASA.</td>
<td>1. This STC is limited to Airbus Helicopters AS 350 B3 post 2017.</td>
<td>15/06/2021</td>
<td>Active</td>
<td>10077284</td>
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<tr>
<td>10077291</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SR11336DS</td>
<td>AS 350 B3</td>
<td>Installation of the Ukraine Border Guard (UBG) collection of systems and equipment specific to the UBG aircraft configuration. The Certification Basis (CB) for the original product remains applicable to this certificate/approval except where amended by additional oralter amendments if indicated on FAA STC SR11336DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of TH...</td>
<td>Master Drawing List 350A45-1927, rev O, dated 12 April 2021; Rotorcraft Flight Manual Supplement 350FMS-577, rev C, dated 3 June 2021; and/or later revisions of the above listed document(s) approved/accepted by EASA.</td>
<td>1. This STC is limited to Airbus Helicopters AS 350 B3 post ECP OP-46922. Refer to approved RFMS for the applicable airspace limitation.3. Prerequisites installation limitations must be complied with: the following are prerequisite installations for this STC: AHI STC SR11333DS, FLIR UltraForce 350-4D4Hi...</td>
<td>16/06/2021</td>
<td>Active</td>
<td>10077291</td>
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<tr>
<td>10077294</td>
<td>DASSAULT AVIATION BUSINESSSERVICES GES SA</td>
<td>FAA STC SR11333DS</td>
<td>FALCON 900EX (F900EX)</td>
<td>TAG-M317 – VARNISH FLIGHT REV 0000006 WIP work of cabin interior wood items finished with varnish</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval except where amended by additional oralter amendments if indicated on FAA STC SR11333DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval without any i...</td>
<td>TAG-M317-MCDL Rev. B dated Sept. 13th 2021; and/or later revisions of the above listed document(s) approved/accepted by EASA.</td>
<td>16/06/2021</td>
<td>Active</td>
<td>10077294</td>
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<td>10077306</td>
<td>SAFRAN CABIN INC.</td>
<td>FAA STC ST02565LA</td>
<td>ERJ 190-100 IGW</td>
<td>Interior Installations and Modifications</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CSS 25.851(a)(6) The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Index List No. J601, Revision A, dated 16 April 2021; and/or later FAA Approved revisions.</td>
<td>01/10/2021</td>
<td>Active</td>
<td>10077306</td>
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<tr>
<td>10077319</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING SERVICES INC.</td>
<td>FAA STC</td>
<td>777-300ER</td>
<td>7777 Cabin Modification for Cargo Transport/Modification of the cabin and removal of existing seats to allow transportation of cargo on the passenger compartment floor.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: Deviation(s): ORI D-130 The following paragraph(s) at a later amendment: CS 25.581(c)(2) Am204The requirements for environmental protection and the associated...</td>
<td>The change is defined with revision 1 of AMES Engineering Order Document EO_M15800102 1, M2, 04112022 The installation has to be accomplished in accordance with revision 6 of AMES Instructions for Continued Airworthiness Document-CA_M15800102 1, M2, 04112022 The aircraft shall be operated in accordance with eq...</td>
<td>21/09/2021</td>
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<td>STC Holder</td>
<td>STC Number</td>
<td>STC/Foreign NAA STC Ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>EMBRAER GFK Ltda. (PC 722)</td>
<td>EMB-136BU</td>
<td>Satcom Aviator 300 Installation</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>- Master Drawings List ref. ESC-MDO-0002 Rev. H dated 6 August 2021.</td>
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<td>24/08/2021</td>
<td>Active</td>
<td>10077347</td>
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<tr>
<td>UAB “FL TECHNICS”</td>
<td>A320-214</td>
<td>Passenger seating arrangement change. This change introduces new passenger seating configuration from current layout (B/C and Y/C150) to Y/C 150 Pax (single passenger economy class layout) to 180 Y/C Pax layout (single passenger economy class layout) on A320 aircraft.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative aviation requirements: - The following paragraph(s) at a later amendment CS25.562 Initial Issue; JAR 25.812 Change 13. The requirements for environmental protection and the associated certified noise and emissions are unchanged and remain applicable to this certificate/ approval.</td>
<td>MDL-S61-01 Issue CAMM Supplement PA-S61-01 initial Issue (MSN 3044); PA-S61-02 Initial Issue (MSN 3601); PA-S61-03 Initial Issue (MSN 3650); for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<td>27/06/2021</td>
<td>Active</td>
<td>10077348</td>
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<tr>
<td>AKV, Inc.</td>
<td>FAA STC SR0275LA</td>
<td>Installation of an AKV, Inc. Data Collector System, K3 PNDCS 1100</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Drawing List No. DCS1100-MDL Rev A, dated 3 April 2021Instructions for Continued Airworthiness, document number DCS1100-ICA Rev. NC (r=n) dated 11 June 2021for later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Pro...</td>
<td></td>
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<td>29/06/2021</td>
<td>Active</td>
<td>10077356</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>STC No.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<th>System Record</th>
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<tbody>
<tr>
<td>1007736</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>60</td>
<td></td>
<td></td>
<td>Mechanical installation of an Iridium antenna on the top fuselage of an aircraft.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact.</td>
<td>Installed in accordance with Service Bulletin Ref: L90_028-12-19 Edition 1 dated 21/11/2021; maintained in accordance with Maintenance Manual Supplement Ref: L90_028-12-30 Edition 2 dated 21/9/2021 or later revisions of the above listed document(s) approved/accepted by the EASA system. This design change only covers the mechanical installation of the antenna. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness...</td>
<td>20/02/2021</td>
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<td>1007752</td>
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<tr>
<td>10077400</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>REV. 1</td>
<td>CL-600-2B19</td>
<td></td>
<td>MMEL Update in CL650: This modification comprises the MMEL update concerning the item installed for the SATCOM system (STC 10074190).</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-MMEL issue 2.</td>
<td>None</td>
<td>18/10/2021</td>
<td>Active</td>
<td>10077403</td>
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<tr>
<td>10077413</td>
<td>MT-PROPELLER ENTWICKLUNG Gmbh</td>
<td>REV. 1</td>
<td>LBA SA141S</td>
<td>SC-7 SERIES 3</td>
<td>Installation of MTV-37-(5) Propellers on Short Skyvan SC-7.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-36 Amendment 59, FAR 36 Amendment 36-37. This certificate/approval invokes a change with an impact on the noise database and/or emissions...</td>
<td>According to MT-Propeller Master Documentation List No. E-3702 Rev. 0: it is to be operated in accordance with MT-Propeller AFM-SP Document No. E-3703 Rev. 0: it is to be installed i.w.o. Installation Instruction Doc. No. E-3705 Rev. 0: it is to be maintained i.w.o. MT-Propeller ICA Document No. E-3706 Rev. for later revisions...</td>
<td>15/11/2021</td>
<td>Active</td>
<td>10077413</td>
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<tr>
<td>10077419</td>
<td>J &amp; C AEROSPACE DESIGN AND CERTIFICATI ON SERVICES EEU Ltd</td>
<td>REV. 1</td>
<td></td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation in Cabin: Revision 1 of 767-300</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-25 Amendment 22, Section 25.861, 25.863, 25.869, 25.1301, 25.1309, 25.1353(c), 25.1800(c), 25.1526a. The requirements for enviro...</td>
<td>MASTER DATA LIST JCAERO/MDL01768 Issue 1.1 or later revisions of the above listed document(s) approved/accepted by the EASA system.</td>
<td>The approval of this design change does not allow the installation of an ELT in open spaces in the cabin.</td>
<td>08/04/2022</td>
<td>Active</td>
<td>10077419</td>
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<tr>
<td>10077422</td>
<td>JET AVIATION AG</td>
<td>747-400</td>
<td></td>
<td></td>
<td>Cabin Reconfiguration Modification of the existing VIP interior by reconfiguring the lower lobe crew rest area into 5 Passenger seats and other cabin installations from a cabin and by implementing specific operational limitations.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: The requirements for enviro...</td>
<td>The change is defined with Revision 9 of the Master Data List: B-747-400: The aircraft shall be operated in accordance with Revision 9 of the Aircraft Flight Manual Supplement UBSL-M-05028AFM-001 or later revisions of the above listed document(s) approved/accepted by the EASA system. - FAA STC ST0301BNY shall be installed prior to or concurrent with this STC...</td>
<td>06/10/2021</td>
<td>Active</td>
<td>10077422</td>
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<tr>
<td>10077432</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING SERVICE ES ING. WALTER STARZACHER Gmbh</td>
<td>767-300</td>
<td></td>
<td></td>
<td>B767 Zero Passenger Configuration The modification reconfigures the Passenger Cabin into a Zero Passenger Configuration by removing the seats and other cabin installations from the cabin and by implementing specific operational limitations.</td>
<td>The Type and OIS Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact...</td>
<td>The change is defined with Revision 0 of the Engineering Order document: B-767 _MT154001158 _ITU 2406/2021 The aircraft shall be operated in accordance with Revision 1 of the Aircraft Flight Manual Supplement document: AF-154001156 _ITU 2 406/2021 The aircraft shall be maintained in accordance with Revision 1 of...</td>
<td>Prior to installation of this change/repair it may be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/10/2021</td>
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<td>10077432</td>
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<td>STC Holder</td>
<td>STC Reference</td>
<td>Aircraft Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>STC Holder</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>ShinMaywa SH21-4</td>
<td>Installation of 23&quot; x 36&quot; Fuselage Cut-out with FS 281 - 321 Internal Cabin Access - Structures Provision Only as per TCCA STC SA21-6</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The certification basis for the original product as amended by TCCA S5-15 Amendment 26: Sections 25.21, 25.25, 25.27, 25.30, 25.32, 25.33, 25.47, 25.56, and 25.61.</td>
<td>PC Power System must not be turned on below 10,000 ft (See AFM Supplement 14517.12-28 Edition 03) Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/10/2021</td>
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<td>STC Holder</td>
<td>JAMET AEROSPACE LTD.</td>
<td>MBB-BK117 D-3</td>
<td>Emergency Floatation System Installation</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The certification basis for the original product as amended by TCCA S5-15 Amendment 26: Sections 25.21, 25.25, 25.27, 25.30, 25.32, 25.33, 25.47, 25.56, and 25.61.</td>
<td>Flight with cover plate installed only. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/06/2023</td>
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<td>STC Holder</td>
<td>ARCHEION HOLDINGS, L.L.C.</td>
<td>DHC-6 SERIES 300</td>
<td>Installation of 23&quot; x 36&quot; Fuselage Cut-out with FS 281 - 321 Internal Cabin Access - Structures Provision Only as per TCCA STC SA21-6</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The certification basis for the original product as amended by TCCA S5-15 Amendment 26: Sections 25.21, 25.25, 25.27, 25.30, 25.32, 25.33, 25.47, 25.56, and 25.61.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/10/2021</td>
<td>Active</td>
<td>10077509</td>
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<td>STC Holder</td>
<td>S-TEC CORPORATION</td>
<td>PC-6/C-H2</td>
<td>Flight with cover plate installed only. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The certification basis for the original product as amended by TCCA S5-15 Amendment 26: Sections 25.21, 25.25, 25.27, 25.30, 25.32, 25.33, 25.47, 25.56, and 25.61.</td>
<td>This STC is applicable to all serial numbers for Pilatus Aircraft, Ltd. Models PC-6B-H2, PC-6B-H2-C, PC-6B-H2-H, PC-6C-H2, PC-6C-H2-H, and PC-6C-H2-P Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/10/2021</td>
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<td>STC Holders</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>DASSAULT AIRCRAFT SERVICES (DAES)</td>
<td>10077548</td>
<td>FALCON 2000/EX</td>
<td>Installation of Satellite Direct (SD) HUB 2.0 WiFi Hub and Wi-Fi antennas</td>
<td>1. Dassault Aircraft Services Master Data List. F250XW-0124A-005 Revision R dated 14 July 2021, or later FAA Approved revision 2.</td>
<td>Aircraft Flight Manual Supplement (AFMS) F250XW-0124A-150, Revision R dated 14 July 2021 or later FAA approved revision supplements.</td>
<td>This STC is applicable to Dassault Aviation Model Falcon 2000/EX airplanes except serial numbers 001 through 005 and 007 through 027. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/10/2021</td>
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<tr>
<td>UMLAUT ENGINEERING GMBH</td>
<td>10077551</td>
<td>BOEING 757-200</td>
<td>INSTALLATION AND CABIN WIFI NETWORK ACTIVATION/This STC installs a Honeywell Jetwave Ka-Band broadband satellite system (including ARINC 791 radome &amp; adapter plate) The structural modifications include:- Installation of internal structural reinforcements-Installation of external...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): Enhanced Airworthiness Programme for Aeroplanes-Systems - ICA on EWS (CRJ) SCH-01 (Equivalent Safety Finding(s)) ESF to CS 25.25(b)(1)(CRF)...</td>
<td>Master Document List ref. UAA-W01-MDL-0001 Issue A-Certification Programme ref. P3- UAA-W01-CPR-0001 Issue D-Compliance Checklist List ref. UAA-W01-CRL-0001 Issue A-Airworthiness Support Manual ref. UAA-W01-AMS-0001 Issue A-Instructions for Continued Airworthiness ref. UAA-W01-ICA-0001 Issue A...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/10/2021</td>
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<tr>
<td>URBE AERO GmbH</td>
<td>10077564</td>
<td>B300</td>
<td>Flight Data Recorder (FDR) upgrade to Universal Avionics FDR-25, P/N 16907-00-00 Extension of Applicability to B300 S/N FL-279</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements...</td>
<td>Universal Engineering Order “FDR Upgrade” document UAA-DOE/DF7, Project 2626, Rev. 02, dated 10.10.2021 Universal Avionics Cockpit Voice Recorder Flight Data Recorder/Installation Manual, document 23-70-08, Rev. 4, dated 30 September 2019 or later revisions of the above listed document(s) approved/ accepted.</td>
<td>Applicable only to Beechcraft B300 S/N FL-246 and S/N FL-279 equipped with Garmin G1000NXi avionics suite Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/10/2022</td>
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<td>BOEING IRELAND LIMITED</td>
<td>10077656</td>
<td>787-8, 787-9, 787-10</td>
<td>Cabin Conversion for On-Seat Cargo</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements...</td>
<td>Change Approval Document Ref. BL-00714-AFP-APP Rev 00 dated 2002 Apr 2021 Certification Programme Ref. BL-00714-AFM02 Rev 2002 Certification Programme Ref. BL-00714-AFM03 Rev 2002 Certification Programme Ref. BL-00714-AFM04 Rev 2002 Certification Programme Ref. BL-00714-AFM05 Rev 2002 Certification Programme Ref...</td>
<td>The aircraft are limited to the carriage of cargo in the passenger cabin economy class passenger seats. Occupancy of the passenger cabin is restricted as defined in the Airplane Flight Manual Supplement, Section 4.1. Cabin crew operating aeroplanes modified by this STC must complete airplane offline...</td>
<td>27/10/2021</td>
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<tr>
<td>SIERRA INDUSTRIES Ltd</td>
<td>10077669</td>
<td>501</td>
<td>Installation of Extended Width Cabin Door</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Branson Aircraft Corporation List of Drawings and Documents, Extended Door, Number 91550, Rev. F, dated 30 July 1981, or later FAA approved revision.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/10/2021</td>
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<td>ARCHEON HOLDINGS, L.L.C.</td>
<td>Master Data List 18H44-MDL Rev. D, dated 30 July 2021 or later FAA STC/AD/MDL Approved revisions, to provide 2 Trimux SATCOM ATS Voice channels, Trimux SATCOM ATS Data Link FANS 1A+ Data channel for Flight Deck use/CPDLC at RCP100 &amp; AOC-5 if RSP180, &amp; 1 optional AOC Voice channel for cabin crew use. For main...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2021</td>
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<td>S4A SOLUTIONS FOR AVIATION, S.L.</td>
<td>Master Data List SR02230AK - Installation of Genesys Aerosystems, Inc. Flight Deck Electronic Flight Instrument System (EFIS)</td>
<td>This design change only covers the installation of the GPS / WAAS antenna and the electrical connection between the antenna and the sensor unit.</td>
<td>28/10/2021</td>
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<td>GENESYS AEROSYSTEMS, INC.</td>
<td>Master Data List 18H44-MDL Rev. D, dated 30 July 2021 or later FAA STC/AD/MDL Approved revisions, to provide 2 Trimux SATCOM ATS Voice channels, Trimux SATCOM ATS Data Link FANS 1A+ Data channel for Flight Deck use/CPDLC at RCP100 &amp; AOC-5 if RSP180, &amp; 1 optional AOC Voice channel for cabin crew use. For main...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/10/2021</td>
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<td>BERINGER AERO</td>
<td>Master Data List 18H44-MDL Rev. D, dated 30 July 2021 or later FAA STC/AD/MDL Approved revisions, to provide 2 Trimux SATCOM ATS Voice channels, Trimux SATCOM ATS Data Link FANS 1A+ Data channel for Flight Deck use/CPDLC at RCP100 &amp; AOC-5 if RSP180, &amp; 1 optional AOC Voice channel for cabin crew use. For main...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/11/2022</td>
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<td>STG AEROSPACE GERMANY GmbH</td>
<td>Master Data List 18H44-MDL Rev. D, dated 30 July 2021 or later FAA STC/AD/MDL Approved revisions, to provide 2 Trimux SATCOM ATS Voice channels, Trimux SATCOM ATS Data Link FANS 1A+ Data channel for Flight Deck use/CPDLC at RCP100 &amp; AOC-5 if RSP180, &amp; 1 optional AOC Voice channel for cabin crew use. For main...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/10/2023</td>
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### List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<th>Associated Technical Documentation</th>
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<th>System Record</th>
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<tr>
<td><strong>AEROTECH &amp; CONCEPT</strong></td>
<td>777-300ER</td>
<td>Cargo in Cabin Installation</td>
<td>The Type and OSL Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>Cargo in Cabin Installation</td>
<td>03/11/2021</td>
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<td><strong>QUIET TECHNOLOGY AEROSPACE, Inc.</strong></td>
<td>GIV-X G450, GIV-X G350</td>
<td>Replacement of the nose cone assembly with a Quiet Technology Aerospace cone assembly.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments</td>
<td>Replacement of the nose cone assembly</td>
<td>04/11/2021</td>
<td>Active</td>
<td>10077605</td>
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<td><strong>QUIET TECHNOLOGY AEROSPACE, Inc.</strong></td>
<td>FAA STC ST04464AT</td>
<td>1. Quiet Technology Aerospace document</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments</td>
<td>1. Quiet Technology Aerospace document</td>
<td>05/11/2021</td>
<td>Active</td>
<td>10077612</td>
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<td><strong>GDS AERO LLC</strong></td>
<td>Falcon 2000, FALCON 2000EX</td>
<td>Installation of GDS Aero Tail Cone APU Firewall Assembly, P/N GF2APU-FW1</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/ approval</td>
<td>Installation of GDS Aero Tail Cone APU Firewall Assembly</td>
<td>05/11/2021</td>
<td>Active</td>
<td>10077615</td>
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<tr>
<td><strong>DOA AESGLOBAL LIMITED</strong></td>
<td>737-800</td>
<td>Introduction of Full Economy Layout</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>Introduction of Full Economy Layout</td>
<td>08/11/2021</td>
<td>Active</td>
<td>10077620</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. Level</td>
<td>Foreign/NAA STC Ref.</td>
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<td>10077665</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>REV. 3</td>
<td>STC04204CH</td>
<td>777-300ER 777-300</td>
<td>2xuc Antenna Installation This project installs an Integral 2xUC Antenna. The Integral 2xUC/Attenuator/Antenna system includes structural provisions and installation of the transmit and receive antennas, an adapter plate and a radome. Revision 1: Transfer of the STC from INTELSAT INFLIGHT LLC to Delta Engineering...</td>
<td>The Type and OSD Certification Baseline for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Findings(s): 14 CFR 25.251(b) – Vibration and Buffeting/Compliance Criteria (Ref Transport Direcotive E02 Mamo ENV/15059-F-1). The requirement...</td>
<td>1. Delta Engineering Master Data List (MDL) D13528-21720 Revision AM 8 September 2023, 2. Instructions for Continued Airworthiness – Electrical Wiring/Interconnect System Instructions, Doc. No. D12434-21720 Rev AC dated 21 July 2023, 3. FAA Approved Aircraft Flight Manual Supplement Doc. No. D12443-...</td>
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<td>10077669</td>
<td>SEATON ENGINEERING CORPORATION</td>
<td>FAA STC 10077669</td>
<td>SEE FAA AML 17/11/2021</td>
<td>10077695</td>
<td>Fabrication and installation of new or replacement of existing instrument panel lights dimmer control.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/ approval.</td>
<td>Seaton Engineering Corp. FAA approved Master Drawing document 9755-002 dated May 10, 2007, or later FAA approved revision.</td>
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<tr>
<td>10077696</td>
<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>GA251192-1-08</td>
<td>WMS Rev. 3</td>
<td>10077695</td>
<td>Installation of Optical System (EOS) EuroFLIR 410 in accordance with Master Data List GVH-11968-01-MDL-01.</td>
<td>The Type and OSD Certification Baseline for aircraft approved by the following additional or alternative airworthiness requirements: Equivalent Safety Findings(s): 14 CFR 25.251(b) – Vibration and Buffeting/Compliance Criteria (Ref Transport Direcotive E02 Mamo ENV/15059-F-1). The requirement...</td>
<td>Rotorcraft Flight Manual Supplement GVH-11968-01-FMS-01 Issue 4 Master Data List GVH-11968-01-MDL-01 Issue 38/19 Change Summary and Declaration of Compliance GVH-11968-01-MDL-01 MOD-0 Issue 2 after revision of the above listed document(s) approved/ accepted/accepted the EASA system</td>
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<td>10077712</td>
<td>CAL STC HOLDINGS LLC</td>
<td>FAA STC 10077712</td>
<td>SEE FAA AML 06/12/2023</td>
<td>10077712</td>
<td>Installation of All LED Cabin Lighting, Cabin and Crew AC Power Outlets, LH Fwd Closet and Misc Cabin Modification on Embraer 135/145. This STC consists in the installation of All LED Cabin Lighting, Cabin and Crew AC Power Outlets, LH Fwd Closet and Misc Cabin Modification on Embraer...</td>
<td>The Type and OSD Certification Baseline for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Findings(s): OR D-01 ‘Emergency Exit Markers allowable’ – ESF. The requirement...</td>
<td>-Master Data List (MDL) E135-MLD-0006, Revision A Aircraft Flight Manual Supplement (AFMS) Document E135- AFMS-0026 Revision IR- Instructions for Continued Airworthiness (ICA) Document E135-ICA-0026 Revision (IR) later revisions of the above listed documents approved/ accepted/accepted the EASA system.</td>
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<td>10077713</td>
<td>APTOZ EHF</td>
<td>AFA STC 10077713</td>
<td>SEE FAA AML 06/12/2023</td>
<td>10077713</td>
<td>Conversion of Main Cabin to Cargo Area</td>
<td>The Type and OSD Certification Baseline for the original product as amended by the following additional or alternative airworthiness requirements: Deviation(s): 0-D0073894 ‘Transportation of cargo in passenger compartments’ the following paragraph(s) is at a later amendment: CS-MEL at Issue 2 for the following...</td>
<td>Instructions for Continued Airworthiness ICA251192-1 Rev C dated 3 November 2018 Supplement AFM251192-1 Rev D dated 3 November 2018 Master Data List MDL305190 Rev B, dated 3 November 2021 Master Data List MDL305190 Rev C dated 3 November 2021 Master Data List MDL305190 Rev D, dated 31 July 2020 Master Data...</td>
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<td>EASA STC No.</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>1007726</td>
<td>SABENA TECHNICS BGC</td>
<td>A330-243 A330-200 CABIN Reconfiguration from 320 PAX (42B/C, 288 Y/C) to 269 PAX (18 B/C, 251 Y/C) with IRE/Reve System Installation</td>
<td>The Type and OSID Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-25.1, 25.29, 25.301, 25.303, 25.305, 25.307, 25.321, 25.337, 25.341, 25.365, 25.561, 25.5s...</td>
<td>Limitations as per §3 of Instructions for Continued Airworthiness as a concurrent requirement of this change, accomplishment of this STC requires the simultaneous accomplishment of the following modification: Installation of a photo-luminescent EFPM as per Luftansa Technik EASASTC No. 10014507 Rev. 0...</td>
<td>18/11/2021</td>
<td>Active</td>
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<td>1007726</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>747-400F LOPA Modification in B747 This design change consists in the replacement of the upper deck seats from aircraft by new ones with similar properties and the same function.</td>
<td>The Type and OSID Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/11/2021</td>
<td>Active</td>
<td>1007726</td>
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<tr>
<td>1007730</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>TCAC SH20-49 ISSUE 2 MBB-Bk117 D-3 MBB-Bk117 D-2 Cabin Heat Outlets Control.</td>
<td>The Type and OSID Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/11/2021</td>
<td>Active</td>
<td>1007730</td>
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<td>1007745</td>
<td>ELLIOTT AVIATION TECHNICAL PRODUCT IDEE LOMPT, LLC</td>
<td>FAA STC SA03120CH B360, B300C B260, B200C 200, 380 Installation of Fuel Flow and Engine Torque DC Transducers</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA03120CH. The requirements for environmental protection and the associated certified noise and/or emissions levels of the...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/11/2021</td>
<td>Active</td>
<td>1007745</td>
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<td>1007746</td>
<td>AVCON INDUSTRIES, Inc.</td>
<td>FAA STC SA04761CH B360C B360 Installation of External Belly Radar POD and Upper SATCOM Radome</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/11/2021</td>
<td>Active</td>
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An agency of the European Union
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<tr>
<th>STC Holder</th>
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<th>Associated Technical Documentation</th>
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<tr>
<td>1007749 AEROTEC &amp; CONCEPT</td>
<td>EMB-145</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements:Special Condition (s): D-01 Medical Evacuation Configuration: the following paragraph(s) at a later amendment: CS25.301, 25.303, 25.305, 25.307, 25.321, 25.331, 25.337, 25.34...</td>
<td>Master Data List AE21075MLD Rev.: 2 dist. 15/11/2021 Aircraft Flight Manual Supplement AE21075AFMS Rev.: 2 dist. 29/10/2021 Instructions for Continued Airworthiness AE21075CA Rev.: 2 dist. 29/10/2021 Changes in the above listed document(s) approved/ accepted under the EASA system.</td>
<td>23/11/2021</td>
<td>Active</td>
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<tr>
<td>1007750 AEROTEC &amp; CONCEPT</td>
<td>DHC-8-402</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: Special Condition (s): D-01 Medical Evacuation Configuration: the following paragraph(s) at a later amendment: CS25.301, 25.303, 25.305, 25.307, 25.321, 25.331, 25.337, 25.34...</td>
<td>Master Data List AE21076MLD Rev.: 2 dist. 15/11/2021 Aircraft Flight Manual Supplement AE21076AFMS Rev.: 2 dist. 29/10/2021 Instructions for Continued Airworthiness AE21076CA Rev.: 2 dist. 29/10/2021 Changes in the above listed document(s) approved/ accepted under the EASA system.</td>
<td>23/11/2021</td>
<td>Active</td>
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<td>1007751 STANDARD AERO BUSINESS AVIATION SERVICE ES, L.L.C.</td>
<td>FAA STC ST0406CH</td>
<td>Installation of a Collins KuSAT-2000 Tail-Mounted SATCOM System and a new access hatch and ladder from Main Deck Cabin to FWD Cargo Compartment</td>
<td>Master Data List (MDL), 1039410, Revision (F), dated 5 November 2021, or later FAA approved revision. Airplane Flight Manual Supplement, 1040118, Revision (A), dated 30 August 2021 (&gt;), or later FAA approved revision is a required part of configuration.</td>
<td>23/11/2021</td>
<td>Active</td>
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<tr>
<td>1007757 SCANDINAVIAN AVIONICS DESIGN A/S</td>
<td>A320-214 A319-133 A319-132</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 art.2 for the following paragraph(s): CS 25.29, CS 25.301, CS25.303, CS 25.305, CS 25.307, CS 25.341, CS 25.561...</td>
<td>AFM Supplement 2120617-AM Issue 4, dated 15 Nov 2021 NEL Supplement 2120617-AM Issue 1, dated 17 Jan 2020 CA Supplement 2120617-CA Issue 1, dated 5 May 2021 Changes in the above listed document(s) approved/accepted under the EASA system.</td>
<td>23/11/2021</td>
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<tr>
<td>1007781 L-3 COMMUNICATIONS INTEGRATED SYSTEMS</td>
<td>FAA STC ST9752SC-D</td>
<td>Installation of an executive interior: This STC consists of the installation of an executive interior with Multiple Rooms and associated equipment and systems in accordance with L-3 Communications Master Data List (MDL) 99010004-000 Rev AM and FAA STC ST9752SC-D</td>
<td>Master Data List (MDL), 99010004-000 Rev AM - Instructions for Continued Airworthiness (ICA) Document 1. <em>ST9752SC-D</em>. M9-V2 - Revision 12. Special Condition (s) <em>ST9752SC-D</em>. MP5D Revision 12. Aircraft Flight Manual Supplement (AFMS) Document referenced STC ST9752SC-D. AFMS Revision 02 changes in the above listed document(s) approved/accepted under the EASA system.</td>
<td>02/12/2021</td>
<td>Active</td>
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<tr>
<td>1007827 UMSAUT ENGINEERING GMbH</td>
<td>747-8</td>
<td>The Type and OSID Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: The Type and OSID Certification Basis for the following paragraph(s) at a later amendment: Appendix 5 of CS25/Amendment 19 paragraph S25.10 (a) &amp; S25.50: The requirements for environmental protection...</td>
<td>FAA STC 79401-0004-000 Rev AM - Instructions for Continued Airworthiness (CCA) Document 1: <em>ST9752SC-D</em>. MMO-V2 - Revision 12. FAA STC 79401-0004-000 Rev AM - Instructions for Continued Airworthiness (CCA) Document 1: <em>ST9752SC-D</em>. MMO-V2 - Revision 12, <em>ST9752SC-D</em>. MMW-V2 - Revision 12, <em>ST9752SC-D</em>. MMW-V2 - Revision 12 changes in the above listed document(s) approved/accepted under the EASA system.</td>
<td>02/12/2021</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
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<td>10077828</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>747-8</td>
<td>10077828</td>
<td>10077834</td>
<td>B747-8 - Cabin Completion Installation of an executive interior including several systems, room–interior components on the main deck. Upper deck is modified interiors of installing telephone and power outlets into the existing interior. The modification covers two configurations for the Private Room...</td>
</tr>
<tr>
<td>10077830</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>747-8</td>
<td>10077830</td>
<td>10077833</td>
<td>B747-8 - SDS Installation Installation of ELBIT Self Defense System (SDS) &quot;C-MUSIC&quot;.</td>
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<td>10077832</td>
<td>AVIONIK STRAUSSING ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td>SA227-CC, SA227-DC</td>
<td>SA227-CC, SA227-DC</td>
<td>Installation of Garmin GA 35 GPS AntennaCase 1 – replacing an existing Trimble GPS antenna Case 2 – new antenna is installed</td>
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<tr>
<td>10077833</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>757-200PF</td>
<td>ST01916NY</td>
<td>ST01916NY</td>
<td>Installation of Dual Rockwell-Collins GSP-4000S, ADS-B (Out); Transponder and Control Panel update</td>
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<tr>
<td>10077834</td>
<td>S-TEC CORPORATION</td>
<td>757-200FF</td>
<td>757-200</td>
<td>757-200</td>
<td>Installation of S-TEC System S5X Two Axis Automatic Flight Control System</td>
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</table>
10077840

**STC Holder**
AIRCRAFT SYSTEMS AND MANUFACTURING G. Inc.

**Models**

**Description**
Installation of an Avionics WiFi System

**Certification basis**
The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: (a) CIS 20 at amendment 4.25.869(a)(11) CIS 29 at amendment 26, 25, 21, 25, 23, 25, 25, 25, 21, 25, 30(1)(x), 25, 30.32,

**Limitations/Conditions**
Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherwise installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

**Date of Issue**
15/12/2021

**Status**
Active

**System Record**
10077633

10077845

**STC Holder**
JET AVIATION AG

**Models**
787-8

**Description**
VVP Cabin Interior Completion

**Certification basis**
The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): D-1101 Injury Criteria for Side-Facing Seats 06 Security Protection of Aircraft Systems and Networks F-093 Non-rechargeable Lithium Bat.

**Limitations/Conditions**
Private Operation (Not-for-Hire, Non-Commercial, Non-Smoking, Non-Max occupancy: 27 Pax, 4 Flight crew, 1 Cabin Crew, Exterior Exit Marking an approved exterior paint livery matching the exterior configuration of the cabin considering the deactivated exits must be applied. Changes to the exterior exit must be applied.

**Date of Issue**
06/12/2021

**Status**
Active

**System Record**
10077645

10077853

**STC Holder**
AIRBUS TRANSPORT INTERNATIONAL SNC

**Models**
A300F4-608ST

**Description**
Special transportation of Airbus Helicopter H225 in NPA configuration (Nippon National Police Agency) Issue 1 of ATI modification ST25MOD002 (applicant’s reference C-132860000500000) (a) certified an removed, reviewed, tested and found compliant with the Certification Basis to deal with non-Airbus sub-assemblies transport. On case-b-

**Certification basis**
The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): D-1101 Injury Criteria for Side-Facing Seats 06 Security Protection of Aircraft Systems and Networks F-093 Non-rechargeable Lithium Bat.

**Limitations/Conditions**
- Change Approval Sheet (CAS) ST25MOD002/06 ST25MOD002/06 Issue 1 dated 18 November 2021- Change Certification Plan (CCP) ST25MOD002/06 Issue 6 – H225/NA Helicopter Transport dated 15 November 2021- Approved Document Dossier (ADD) ST25MOD002/06 ADD Issue 3 including AFM and WBM supplements as enclosed in Appendix

**Date of Issue**
07/12/2021

**Status**
Active

**System Record**
10077653

10077859

**STC Holder**
ROCKWELL COLLINS, Inc.

**Models**
FAA STC ST01891AT

**Description**
Installation of Collins VHF-700B/900B Transceiver and Collins CMU-900ACARS

**Certification basis**
The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.

**Limitations/Conditions**
Collins Master Data List 967014/005, Revision AG dated 1 June 2005, or earlier FAA approved revision Airplane Flight Manual Supplement AMS Issue 6, dated 23 November 2000, or earlier FAA approved revision.

**Date of Issue**
08/12/2021

**Status**
Active

**System Record**
10077659

10077861

**STC Holder**
SABENA TECHNICS BOD

**Models**
REV. 2

**Description**
Cabin modification Reconfiguration of the cabin area (S-26-2420) and communication center ‘Com Gouv’ Mod (S-26-2422) Rev. 1 type corrected. Rev. 2: removal of OSD limitation

**Certification basis**
The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Deviation(s): CR D-132 ‘Exits with non-adjacent Cabin Attendant Seat issue’ (the following paragraph(s) at a later amendment: JAR-25 at change amendment 14:25.1423(11).

**Limitations/Conditions**
- The pre modification for this configuration is covered by EASA STC 10032391
- Aeroclasses modified per this configuration are determined to be airworthiness ‘variant’ to the previous changes.
- Nevertheless, for Cabin Crew who have not undergone training satisfying the approval of previous STCs, the overall-

**Date of Issue**
19/10/2022

**Status**
Active

**System Record**
10077861
<table>
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<tr>
<th>STC Holder</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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</thead>
<tbody>
<tr>
<td>BOMBARDIER Inc.</td>
<td>Demonstration of TPEO Tolerance/ This modification provides data to support</td>
<td>The Type and OIS Certification Bases (CB) for the original product and the</td>
<td>- Master Data List, Aerocore document No. 40211-01-A-MDL-R00. Instructions for continued</td>
<td>This STC does not constitute an operational approval for the use of portable electronic</td>
<td>10/12/2021</td>
<td>Active</td>
<td>10077918</td>
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<tr>
<td></td>
<td>operational approval of reassembling portables electronic devices (TPEO).</td>
<td>associated certified noise and/ or emissions levels of the product are unchanged and</td>
<td>environmental protection and the associated certified noise and/ or emissions levels of the product</td>
<td>devices (PED) on board. Operational approval must be obtained by the responsible</td>
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<td>The data can also be viewed in support of installation/activation approval of</td>
<td>remain applicable to this certificate/ approval w/...</td>
<td>remain applicable to this certificate/ approval w/...</td>
<td>airworthiness authority. This STC does not constitute installation approval of</td>
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<td></td>
<td>wireless access point or other radiofrequency system using wireless d...</td>
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<td>a wireless radiofrequency system using wireless data...</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>A330-300 CABIN Reconfiguration from 285 PAX (30 B/C, 265 Y/C) to 387 PAX (387 Y/C)</td>
<td>The Type and OIS Certification Bases (CB) for the original product as amended by the</td>
<td>Instructions for continued airworthiness, Aerocore document No. 40211-01-A-ICA-R00 later</td>
<td>Limitations as per §3 of Instructions for Continued Airworthiness as a concurrent</td>
<td>13/12/2021</td>
<td>Active</td>
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<tr>
<td>FAA</td>
<td>Installation of Garmin Aerosystems Electronic Flight Instrument System (EFIS) and associated equipment</td>
<td>the following additional or alternative airworthiness requirements the following</td>
<td>revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>requirement of this change, accomplishment of this STC requires the simultaneous</td>
<td></td>
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<td>GENESISYS</td>
<td>The Certification Basis for the original product remains applicable to the</td>
<td>paragraph(s) at a later endorsement: CS-25 at amendment 4-25.1, 25.29, 25.301, 25.303, 25.305, 25.307, 25.321, 25.327, 25.341, 25.365, 25.361, 25.365, 25.375...</td>
<td></td>
<td>accomplishment of the following modification: Installation of a photo-luminescent EFPM as</td>
<td></td>
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<tr>
<td>A319-133</td>
<td>FAA</td>
<td>SR01673SE</td>
<td>- Master Data List (MDL) 651-045049, Revision G, dated 21 April 2021- Rotorcraft Flight Manual Supplement (RFSMS) 17-673FMS-0490 and/ or 17-673FMS-0480. Revision A dated 04 August 2021 Rotorcraft Instructions for Continued Airworthiness (RICA) 17-12004, Revision A, dated 02 April 2021 and 17-12005, Revision A</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>13/12/2021</td>
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<td>10077926</td>
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<tr>
<td>LUFTHANSA TECHNIK</td>
<td>The Type and OIS Certification Bases (CB) for the original product as amended by</td>
<td>the following additional or alternative airworthiness requirements the following</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse</td>
<td>13/12/2021</td>
<td>Active</td>
<td>10077930</td>
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<td>GARMIN INTERNATIONAL, Inc.</td>
<td>Installation of Garmin GFC 600H Flight Control System Rev. 1: Approval reference for OIS-FCD and removal of associated documentation on EASA certificate.</td>
<td>the following additional or alternative airworthiness requirements. - Special Condition(s) adopted FAA 27-043 SC (System Safety). - The following paragraph(s) at a later endorsement: In addition to the STC Certification Basis rep.</td>
<td></td>
<td>Prior to the installation of this change/repair/ any other modification/installation of the</td>
<td>20/05/2022</td>
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<td>10077931</td>
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Note: The document contains a table listing various supplemental type certificates with their respective details.
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<th>Certification basis</th>
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<tr>
<td>T.P.S. S.P.A.</td>
<td>Baskets installation on AS350 series</td>
<td>Master Document List TP3530-1200- M4-00/Rotorcraft Flight Manual Supplement</td>
<td>Prior to installation of this change/repair it must be determined if there be an adverse effect upon the airworthiness of the product.</td>
<td>14/12/2021</td>
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<td>TP3530-1200- M4-00/Instructions for Continued Airworthiness TP3530-1200- M4-00/ later revisions of the above listed documented() approved/accepted/prior to the EASA system.</td>
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<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
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<td>AS350-243</td>
<td>Passenger Seats Rearmangement/Passenger seats rearrangement in the AFT cabin. It includes the placement of five Y/C passenger seats by three double passenger seat model 6810 by Safran, to be used as Crew Rest Area when required.</td>
<td>MDL Passenger Seats Rearrangement in Airbus AS330 Ref. 330-251-18-02 Issue 3 of 15/12/2021. FMS Passenger Seats Rearmangement in Airbus AS330 Ref. 330-251-12-28 Issue 1 of 15-12-2021 or later revisions of the above listed documented() approved/accepted/prior to the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined if there be an adverse effect upon the airworthiness of the product.</td>
<td>16/12/2021</td>
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<td>FAO ST04306AT-D</td>
<td>Installation of Tailwind 500 Satellite TV System in accordance with Gulfstream Aerospace Corporation Index List as referenced on FAA Approved Model List (AML) GC73507D003 Rev. 1 corrects typos.</td>
<td>Faa Approved Model List (AML) GC73507D003, Revision A, dated 08 December 2021.Refer to the AML for additional limitations and conditions as applicable FAA / ODA accepted Instructions for Continued Airworthiness/FAA / EAR approved EWSI instructions for Continued Airworthiness/FAA / ODA approved AirlineO</td>
<td>This STC is valid for aircraft types and models approved by EASA and the FAA and the FAA/AML GC73507D003 Revision A, dated 08 December 2021. Prior to installation of this change/repair it must be determined if there be an adverse effect upon the airworthiness of the product.</td>
<td>21/12/2021</td>
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<td>STC Holder 1</td>
<td>STC Type 1</td>
<td>STC File 1</td>
<td>Description 1</td>
<td>Certification Basis 1</td>
<td>Associated Technical Documentation 1</td>
<td>Limitations/Conditions 1</td>
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<td>STC Holder 2</td>
<td>STC Type 2</td>
<td>STC File 2</td>
<td>Description 2</td>
<td>Certification Basis 2</td>
<td>Associated Technical Documentation 2</td>
<td>Limitations/Conditions 2</td>
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<td>STC Holder 3</td>
<td>STC Type 3</td>
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<td>Description 3</td>
<td>Certification Basis 3</td>
<td>Associated Technical Documentation 3</td>
<td>Limitations/Conditions 3</td>
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Note: The table above represents a sample list of EASA Supplemental Type Certificates. The actual list can be found on the EASA website at easa.europa.eu.
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<th>STC Record</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>10078015</td>
<td>ALPINE AEROTECH LP</td>
<td>212, 412, 412E5, 205A-1</td>
<td>Installation of Garmin GTX 335 ADS-B Transponder</td>
<td>The Type and OSD Certification Bases (CB) for the original products/aircraft applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval w...</td>
<td>Master Data List AAL-280-084-001 Rev. B, TCCA-approved 25 November 2021; Rotorcraft Flight Manual Supplement, AAL-280-085-401 Rev. A, TCCA-approved 25 November 2021.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2021</td>
<td>Active</td>
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<tr>
<td>10078016</td>
<td>BOGGI AERONAUTICS S.R.L.</td>
<td>AS 350 B, AS 350 BB AS 350 B3, AS 350 BA, AS 350 D AS 350 B1, AS 350 B2</td>
<td>External Mirror System 350P025 installed on RHS and/or LHS as described in the Configuration Control Document 350P025-20-CDD.</td>
<td>The Type and OSD Certification Bases (CB) for the original products/aircraft applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval w...</td>
<td>Configuration Control Document 350P025-20-CDD, Issue 1, Master Document List 350P025-20-M-00, Issue 2, RHS and/or LHS heated and movable mirror, RHS and/or LHS not electrically controlled Flight Manual Supplement 350P025-20-FMS-10, Instruction/Instructions for Continued Airworthiness 350P025-20-ICA-10,R...</td>
<td>This STC includes the external mirror configuration previously approved EASA STC 16048606. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/12/2021</td>
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<td>10078016</td>
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<tr>
<td>10078017</td>
<td>HAVACLİK MUHENDİSLİK VE SERTİFİKAŞİON ETİBER SANAYİ VE TİCARET ANONİMŞIRKETİ</td>
<td>A340-542</td>
<td>Installation of VIP interior furnishings/Four interior dressing units installed in an existing meeting room.</td>
<td>The Type and OSD Certification Bases (CB) for the original products/aircraft applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval w...</td>
<td>Master Document List, MDL-70150, Issue 0, dated 15th December 2021; Airplane Flight Manual Supplement, TS1150R50, Issue 0, dated 14th December 2021; later revisions of the above listed documents(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/12/2021</td>
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<td>10078017</td>
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<td>10078020</td>
<td>SPAES GmbH &amp; Co. KG</td>
<td>429</td>
<td>Aocam, Tetra, GPS Antenna Installation</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: C5-25 Amrt 25_25.813(a)(ii)(f), 25.1726. The requirements for environmental protection and the associated/certified noise an...</td>
<td>Instruction for Continued Airworthiness ICA-429-461-29-01 Rev. 1.1 dated 03/11/2021; Flight Manual Supplement FMS-429-461-23-01 Rev. 1.0 dated 17/09/2021; Master Document List MDL-429-461-23-01 Rev. 1.7 dated 17/12/2021; later revisions of the above listed documents(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/12/2021</td>
<td>Active</td>
<td>10078020</td>
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<tr>
<td>10078021</td>
<td>BOEING IRELAND LIMITED</td>
<td>REV. 2</td>
<td>Deactivation and removal of GoGo ABS Ku-Band Antenna (FAA STC/STC/105111NY), including removal of all structural provisions, filling the vacant/void/ vacant/void area and or location of skin replacement in Section 48/Rev. 1 corrects the Limit...</td>
<td>The Type and OSD Certification Bases (CB) for the original products/aircraft applicable to this certificate/ approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval w...</td>
<td>Master Data List provided in Change/Repair Approval Document, Doc.BIL-00204-ICAO Rev. 0, Instructions for Continued Airworthiness, Doc. BIL-00204-ICAO Rev. 0 for Continued Airworthiness, Doc. BIL-00204-ICAO Rev. 0 for Continued Airworthiness, Doc. BIL-00204-ICAO Rev. 0 for Continued Airworthiness, Doc. BIL-00204-ICAO Rev. 0 for Continued Airworthiness.</td>
<td>Aircraft are to be maintained in accordance with Boeing Ireland Limited Instructions as detailed below. Airplanes modified by this STC must be maintained in accordance with Instructions for Continued Airworthiness, Doc. BIL-00204-ICAO Rev. 0 and AMM/AMM-350R50.</td>
<td>29/04/2024</td>
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<td>10078021</td>
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<td>STC Holder</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10078027</td>
<td>LUTFTHANSTECN MICHAEL ENDER</td>
<td>REV. 1</td>
<td></td>
<td>TAPF UN17E</td>
<td>retrofit of Rotax 914F3 engine with MTV-21-AC-FCF175-05 Propeller, TME- STC-259-1/2018</td>
<td>the type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement 3000035-AM I, issue 3., Flight Manual Supplement 300035-AM II, issue 1., Certification Documents List 3000035 DOC, issue 3., Drawing List 3000035-DL, issue 3., Instructions for Continued Airworthiness 2019011 rev0, 2019011 rev1, 2019011 rev2</td>
<td>Prior to installation of this change/repair it must be determined that the internation relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/05/2022</td>
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<td>10078032</td>
<td>SWISS HELICOPTER ENGINEERING AG</td>
<td></td>
<td></td>
<td>KA-32A11BC</td>
<td>Installation of an additional bubble window on the right-hand side installation of a bubble window with enhanced curvature for the right-hand of a Kamov KA-32A11BC, which is already equipped with a small rectangular bubble window as original installed part.</td>
<td>the type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement 3000035-AM I, issue 3., Flight Manual Supplement 300035-AM II, issue 1., Certification Documents List 3000035 DOC, issue 3., Drawing List 3000035-DL, issue 3., Instructions for Continued Airworthiness 2019011 rev0, 2019011 rev1, 2019011 rev2</td>
<td>Prior to installation of this change/repair it must be determined that the internation relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/12/2021</td>
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<td>10078041</td>
<td>SCANDINAVIAN AVIONICS DESIGN AS</td>
<td>REV. 1</td>
<td></td>
<td>AW139</td>
<td>RDR-7000 Weather Radar installation The RDR-7000 Weather Radar installation is a replacement of the existing PRIMUS 70s and PRIMUS 660 weather radars installed on the AW139 helicopters Long Noise configuration.</td>
<td>the certification basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement 3000035-AM I, issue 3., Flight Manual Supplement 300035-AM II, issue 1., Certification Documents List 3000035 DOC, issue 3., Drawing List 3000035-DL, issue 3., Instructions for Continued Airworthiness 2019011 rev0, 2019011 rev1, 2019011 rev2</td>
<td>Prior to installation of this change/repair it must be determined that the internation relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10078052</td>
<td>MEDITERRANEAN AVIATION COMPANY Ltd. (MEDAVIA)</td>
<td></td>
<td></td>
<td>ERU 190-100 ECJ</td>
<td>Certification of Alternative Life Raft on Lineage 1000</td>
<td>the certification basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement 3000035-AM I, issue 3., Flight Manual Supplement 300035-AM II, issue 1., Certification Documents List 3000035 DOC, issue 3., Drawing List 3000035-DL, issue 3., Instructions for Continued Airworthiness 2019011 rev0, 2019011 rev1, 2019011 rev2</td>
<td>Prior to installation of this change/repair it must be determined that the internation relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/01/2022</td>
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<td>10078055</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>REV. 1</td>
<td></td>
<td>EMB-135LR, EMB-145LR</td>
<td>IRIDEM ANTENNA INSTALLATION REV 1: Maintenance interval inspections update</td>
<td>the type and OSD Certification Bases for the original product remain applicable to the following additional or alternative airworthiness requirements: Special Condition (s): Non-rechargeable Lithium Battery Installations. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>Flight Manual Supplement 3000035-AM I, issue 3., Flight Manual Supplement 300035-AM II, issue 1., Certification Documents List 3000035 DOC, issue 3., Drawing List 3000035-DL, issue 3., Instructions for Continued Airworthiness 2019011 rev0, 2019011 rev1, 2019011 rev2</td>
<td>Prior to installation of this change/repair it must be determined that the internation relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/08/2022</td>
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<td>10078064</td>
<td>LUNA TECHNOLOGIES LLC</td>
<td></td>
<td></td>
<td>FAA STC ST07222SE</td>
<td>Installation of LT-4098 series Integrated LED Display Panels on the associated LT-2201 series LED Pushbutton Switches</td>
<td>the certification basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if mandated on FAA STC ST07222SE. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>1. Master Data List (MDL) MDL-4098-001, Revision A, dated 20 October 2020, and later FAA-approved revision 2. Instructions for Continued Airworthiness (ICA) ICA-4098-001, Revision A, dated 29 July 2019, and/or ICA-2201-001 (s), revision 2, dated 29 July 2019 as applicable, or later FAA-accepted rev.</td>
<td>Prior to installation of this change/repair it must be determined that the internation relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/01/2022</td>
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<td>10078067</td>
<td>LEARJET Inc.</td>
<td>Installation of wiring provisions and structural provisions for a Wing Collision Avoidance System</td>
<td>The Certification Basis for the original product remains applicable to the aircraft certificated noise and/or emissions levels of the product.</td>
<td>Master Drawing UTI GC31005011, revision B dated 10/03/01 or later FAA approved revision.</td>
<td>01/01/2022</td>
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<td>10078070</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>Modification of Rockwell Collins Radio Interface Unit (RIU), SXM/Repeater and Pro Line Fusion Display System Software</td>
<td>The Certification Basis for the original product remains applicable to the aircraft certificated noise and/or emissions levels of the product.</td>
<td>Aircraft affected by Airworthiness Directive 2021-05-17 must be equipped with software specified in Rockwell Collins Service Information Letter FDS A-650-19-1 prior to installation of this type design change. For model specific limitations, see Model List (AML) R-671, Rev. (-), or later approved revisions.</td>
<td>06/01/2022</td>
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<td>10078131</td>
<td>ASI INNOVATION</td>
<td>Installation of multipurpose “POD SC2” on the LH multifunction arm mount P/N RS83-001-12100-00.</td>
<td>The Certification Basis for the original product remains applicable to the aircraft certificated noise and/or emissions levels of the product.</td>
<td>The multipurpose POD SC2 installation is certified as a provisioning for ageneric payload only. Any further installation inside or outside the podshell require a specific airworthiness approval. The multipurpose POD SC2 installation is only certified for Category B and/or night VFR operations.</td>
<td>13/01/2022</td>
<td>Active</td>
<td>10078131</td>
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<td>10078133</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td>Installation of Wi-Fi Viasat installation</td>
<td>The Certification Basis for the original product remains applicable to the aircraft certificated noise and/or emissions levels of the product.</td>
<td>As per the AFM Supplement AFMS-NA0635 Rev. 1, the installation of this equipment must be evaluated to assure satisfactory compliance with the applicable airworthiness standards. Prior to installation of this change/repair it must be determined if the equipment is certified.</td>
<td>13/01/2022</td>
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<td>10078133</td>
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<td>10078142</td>
<td>TEAM J.A.S., Inc. dba PMA DESIGN AND MANUFACTURING</td>
<td>Installation of a PMA Design and Manufacturing nose landing gear (NLG) shock strut assembly. DGC-6 SERIES 400 DGC-6 SERIES 300 DGC-6 SERIES 200 DGC-6 SERIES 100.</td>
<td>The Certification Basis for the original product remains applicable to the aircraft certificated noise and/or emissions levels of the product.</td>
<td>Prior to installation of the change/repair it must be determined if the equipment is certified.</td>
<td>14/01/2022</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>10078145</td>
<td>HELILONE (NORWAY) AS</td>
<td>REV. 1</td>
<td>AS 365 N3</td>
<td>Design change AHSE-72010, installing a CMA-9000 FMS and dual CMA-320XN/GSU GPS receivers and relevant antennas on AS 365 N3 helicopters allowing the following: UNAV capability, RNP0.1 for enroute/continental, RNP0.3 for enroute/contemporary, (STAR) and...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following: additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: Elect to comply to CS-ACNS ACNS.C.PBN 190 to ACNS.C.PBN 2150, Issue 2. The requirements for environmental pro...</td>
<td>Dept AHSE-72010, &quot;CMA-9000 FMS Installation Master Drawing/Document no. AHSE-72010 “Certification Programme”, Revision B, dated 02/12/2021.</td>
<td>&quot;installation...</td>
<td>19/01/2022</td>
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<td>10078158</td>
<td>ACC COLUMBIA JET SERVICE GMBH</td>
<td>REV. 5</td>
<td>A340-3I1/312-313 A330-322I/323-323</td>
<td>Installation of In-Service Aircraft for a Global Observing System (IAGOS) P2 in A330 and A340 aeroplanes – Package P2b; 1 documentation updated as per minor change approval8MCA_P2b_22-91267, rev0/Rev.2 documentation updated as per minor change approval8MCA_P2b_23-90402_rev0Rev.3 correction of...</td>
<td>The Type and OSD Certification Bases (CB) for the original product underwent changes applicable to this certification approval. The requirements for additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-27 Amendment 2b, 27.605(c), (d) and CS-27 Amendment 2b, 27.619(a), (b), (c), (d) and CS-27 Amendment 2b, 27.619(b), (c), (d) and...</td>
<td>Master Document List MDOLC_P2b_24-90059 MDOOLC_MDOLC_34-90059 rev0/Master Document List MDOLC_P2b_24-90059 MDOOLC_MDOLC_34-90059 rev0/Master Document List MDOLP_P2b_21-90011 rev 3/Instructions for Continued Airworthiness ICA_P2b_21-90011_rev 3/rev0/rev1 later revisions of the above listed document(s) approved.</td>
<td>Pre-condition of the installation of this design change is...</td>
<td>17/04/2024</td>
<td>Active</td>
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<tr>
<td>10078169</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>EC135 T3</td>
<td>Powerline Mission Equipment and Load Carrying Component(s) Issue/Addition of Type 4b and 4c Workstation Basket for static powerline operations (basket deposited on powerlines) to the STC 10065979 (Rev 0) previously approved under EASA airframe revision 1/Introduction of revised fatigue life limits...</td>
<td>The Type and OSD Certification Bases (CB) for the original product underwent changes applicable to this certification approval. The requirements for additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-27 Amendment 2b, 27.605(c), (d) and 27.619(b), (c), (d) and 27.1585(c) CS-27 and...</td>
<td>Initial Issue/Design Compliance Reports: Certification Programme DCP13535034-004-00 Initial Data Definition:- Master Record Index MRU: 13530602-02 Issue A, Type 4b-400 Basket: Master Record Index MRU: 13530602-02 Issue A, Type 4b-600 Basket: Master Record Index MRU: 13530602-02 Issue B, 4c Basket: Master Index MRU: 13530602-02 Issue B, 4c Basket: Master Index MRU: 13530602-02 Issue B, 4c Basket: Master Index MRU: 13530602-02 Issue B, 4c Basket: Master Index MRU: 13530602-02 Issue B, 4c Basket: Master Index MRU: 13530602-02 Issue B, 4c Basket: Master Index MRU: 13530602-02 Issue B, 4c Basket: Master Index MRU: 13530602-02 Issue B, 4c Basket: Master Index M...</td>
<td>Limited to helicopters with CPDLS output variant/Prior to installation of this change/repair it must be determined that no...</td>
<td>28/06/2022</td>
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<tr>
<td>10078181</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 2</td>
<td>FAA ST04333AT-D</td>
<td>GIV, GIV-X, GV, GV-SP</td>
<td>Installation of Salcom Direct Gateway and WiFi Hub System on Gulfstream GIV, GV, GIV-X and GV-SP model aircraft. Rev. 1 corrects Technical Documentation referenced. Rev. 2 corrects Technical Documentation reference.</td>
<td>The Type and OSD Certification Bases (CB) for the original product underwent changes applicable to this certification approval. The requirements for additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-27 Amendment 2b, 27.605(c), (d) and 27.619(b), (c), (d) and 27.1585(c) CS-27 and...</td>
<td>• Supplemental Type Certificate (STC) ST04333AT-D-I, Installation of Salcom Direct Gateway and WiFi Hub System, Date Approved September 12, 2020 • Index List ref. CE12028851 Revision A, dated October 12, 2020 • Airplane Flight Manual Supplement CE12028851, Revision 2, dated June 8, 2020 for Gulfstream IV...</td>
<td>Prior to installation of this change/repair it must be determined that no...</td>
<td>03/02/2022</td>
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<tr>
<td>10078182</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>REV. 2</td>
<td>STC-AT-150 DHC-6 MX-25 Integration N L3 Wesmac MX-25 camera is camera in the nose of the aircraft. The Voss-tech...</td>
<td>The Certification Basis for the original product remains applicable both/this certification approval. The requirements for environmental protection and the associated collateral noise and emissions levels of the product are unchanged remain applicable to this certification approval without any impact...</td>
<td>Master Data List, Change Approval STC-AT-150, Document Number STC-150-G-AS, Rev. Off-Flight Manual Supplement for DHC-6 Series 400, MX25 Integration/Document Number STC-150-G-APM, Rev./Maintenance Manual Supplement for DHC-6 Series 400, STC-AT-150 DHC-6MOS25 Integration, Document Number STC-150-G-AMM, Rev...</td>
<td>Prior to installation of this change/repair it must be determined that...</td>
<td>21/01/2022</td>
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<td>10078190</td>
<td>AVIATION SPECIALTIES UNLIMITED, INC</td>
<td>FAA S016345E</td>
<td>AB412, AB412 EP 212, 412, 412EP</td>
<td>EASA Validation of the Night Vision Imaging Systems (NVIS) Supplemental Type Certificate (STC) SR016345E</td>
<td>The Type and OSD Certification Bases (CB) for the original product changes applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Project Specific Certification Plan (PSCP) ref. ASU-PSCP: AB412-2586: Revision IR, dated 01/06/2021.</td>
<td>Rotorcraft Flight Manual (RFM) Supplement ref. FMS-41200-4 Revision J, dated 23/06/2021 and approved by the FAA on 23/01/2022.</td>
<td>26/01/2022</td>
<td>Active</td>
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<tr>
<td>10078191</td>
<td>3S CERTIFICATION, LLC</td>
<td>FAA STC SA00028MC</td>
<td>PC-12/47E</td>
<td>Installation of non-essential antennas and radomes.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA00028MC. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>1. Sierra Nevada Corporation (SNC), Master Drawing List (MDL) 0126-00-9305, Rev A, dated 08 July 2020 or later FAA approved revision J. For configurations 01260000-01 through -04 defined in design datalist MDL-01260001, the following manuals apply: Airplane Flight Manual, Supplement (AFMS), SNC document No.</td>
<td>The approval is limited to aircraft serial numbers 1186 and 2126. This STC is intended for military, public or government use. Aircraft only. Installation of this system on aircraft for commercial use is prohibited. Prior to installation of this change/repair it must be determined that the interrelations...</td>
<td>26/01/2022</td>
<td>Active</td>
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<tr>
<td>10078216</td>
<td>SIERRA NEVADA CORPORATION</td>
<td>FAA STC SA00030MC</td>
<td>PC-12/47E</td>
<td>Installation of non-essential radio rack, communications equipment antennas</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA00030MC. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>1. Sierra Nevada Corporation (SNC), Master Drawing List, Document Number MDL-03030001, Revision E, dated 18 April 2012. FAA approved 11 July 2013 or later FAA approved revision J. For configurations MDL-03030001-01 through -04 defined in design datalist MDL-03030001-01 the following manuals apply: Airplane Flight Manual Supplement (AFMS), SNC document No.</td>
<td>The approval is limited to aircraft serial numbers 1186 and 2126. This STC is intended for military, public or government use. Aircraft only. Installation of this system on aircraft for commercial use is prohibited. Prior to installation of this change/repair it must be determined that the interrelations...</td>
<td>01/02/2022</td>
<td>Active</td>
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<td>10078217</td>
<td>SIERRA NEVADA CORPORATION</td>
<td>FAA STC SA0007MC</td>
<td>PC-12/47E</td>
<td>Installation of antennas and provisions to mount antennas</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval, except where amended by additional or later amendments if indicated on FAA STC SA0007MC. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>1. Sierra Nevada Corporation (SNC) Master Drawing List, Document Number MDL-01260001, Revision D, dated 14 April 2012 or later FAA approved revision J.</td>
<td>The approval is limited to aircraft serial numbers 1186 and 2126. Prior to installation of this change/repair it must be determined that the interrelations...</td>
<td>01/02/2022</td>
<td>Active</td>
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<tr>
<td>10078220</td>
<td>J &amp; C AEROSPACE DESIGN ANDCERTIFICATI ON SERVICES EUU Ltd.</td>
<td>737-900 737-800, 737-700, 737-600</td>
<td>Integration of cargo seat bags for transportation of cargo in Boeing 737 &quot;CLASSIC&quot; AND &quot;NEXT GENERATION&quot; aircrafts.</td>
<td>The Type and OSD Certification Bases (CB) for the original product include the following additional or alternative airworthiness requirements. Derivation(s): CRI D-01. Transport of cargo in passenger compartments. The requirements for environmental protection and the associated certified noise are unchanged.</td>
<td>J &amp; C Aerospace Master Data List JCAERO/MIL/ 0689 J &amp; C Aerospace Aircraft Flight Manual Supplement JCAERO/AFM/ 0689-1 J &amp; C Aerospace Aircraft Maintenance Manual JCAERO/AMM/ 0689-1.</td>
<td>The allowance of transport cargo in passenger compartment is limited to 000 flight hours following installation of this change/refurbishment to 31 July 2022, whatever occurs first.</td>
<td>02/02/2022</td>
<td>Active</td>
<td>10078220</td>
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<tr>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC no.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>INNOVATIVE SOLUTIONS AND SUPPORT, Inc.</td>
<td>FAA STC ST003680B</td>
<td>767-300F 767-300 767-200</td>
<td>Installation of an Innovative Solutions &amp; Support, Inc. (IS) S/N&lt;-&gt;&lt;S&gt; Integrated Standby Unit (ISU), Part Number 9D-80190-7, loaded with top level software TH-80190-8 as defined in section Limitations/Conditions (or later approved top level software)</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-25 and/or 14(a) of 25.1529, 25.3570(a), 25.366(a)(b)(c)(d), 25.167(a)(b)(c)(d)(e), 25.168(a)(b)(c)(d), 25.1529, 25.154(a)(b)(c), CS-25 and/or 14(a) of CS-25 and/or 14(a) of 25.1529, 25.3570(a), 25.366(a)(b)(c)(d), 25.167(a)(b)(c)(d), 25.168(a)(b)(c)(d), 25.1529, 25.154(a)(b)(c)</td>
<td>ISD-80190-Master Data List (MLD) 1D-12292, Revision 1, dated 18 March 2024. (Note: S/N&lt;-&gt;&lt;S&gt;) &gt;6 Aircraft Flight Manual Supplement (AFMS) 1D-12385, Revision R, FAA approved on 21 March 2019. (Note: S/N&lt;-&gt;&lt;S&gt;) Instractions for Continued Airworthiness (ICA) 1D-12370, Revision 1, dated 2 September 2019. (Note: S/N&lt;-&gt;&lt;S&gt;) EASA STC no. 9846-1-1</td>
<td>21/02/2022</td>
<td>Active</td>
<td>10872829</td>
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<tr>
<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td></td>
<td>777-200</td>
<td>IFE Monitor Replacement</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-25 and/or 14(a) of 25.1529, 25.3570(a), 25.366(a)(b)(c)(d), 25.167(a)(b)(c)(d), 25.168(a)(b)(c)(d), 25.1529, 25.154(a)(b)(c)</td>
<td>Master Drawing List N0674-D10 Rev RI0 Rev 1C Certification summary N0674-D90 Rev 1C Roger replicas of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>08/02/2022</td>
<td>Active</td>
<td>10872870</td>
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<tr>
<td>SET FLUGTECHNIK GmbH</td>
<td>FAA STC SA0282NY</td>
<td>DHC-6 SERIES 400 DHC-6 SERIES 300 DHC-6 SERIES 200 DHC-6 SERIES 100</td>
<td>Replacement Cargo Door latch/ing mechanism on the Viking DHC-6 Series aircraft</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-25 and/or 14(a) of 25.1529, 25.3570(a), 25.366(a)(b)(c)(d), 25.167(a)(b)(c)(d), 25.168(a)(b)(c)(d), 25.1529, 25.154(a)(b)(c)</td>
<td>Document Control List, PDM-MDL0013, Rev B, dated 19 October 2021, contains FAA approved revisions. Instructions for Continued Airworthiness (ICA), PDM-ICA013, Rev A, dated 31 July 2021 or later FAA approved revisions.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2022</td>
<td>Active</td>
<td>10872821</td>
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<td>STC Holder</td>
<td>STC Type</td>
<td>STC Ref.</td>
<td>Rev.</td>
<td>Foreign/NAA STC</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>NEW-SENSE GEOPHYSICS LIMITED</td>
<td>STC</td>
<td>10078614</td>
<td>1</td>
<td>TCCA SH07-44</td>
<td>AS 350 B, AS 350 B1, AS 350 B2, AS 350 B3, AS 350 BA, AS 350 D</td>
<td>Fibrotex stinger installation on AS 350 Rev. 1 corrects a typo in Associated Technical Documentation.</td>
<td>The Type and OSD Certification Bases (CB) for the original product. Additional requirements for the certified/approval are applicable to the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document MOE-ATS-684-MDL-1000 Revision NC, dated 29 April 2008; Installation Instructions III-A350-FIG-1000, Revision NC, dated 17 September 2007; Maintenance Manual ATS-684-MDL-1000 Revision NC, dated 28 April 2008; Rotorkraft Flight Manual Supplement AAC- RFMS AS 350 FIG-1001, Issue 01, dated 28C...</td>
<td>09/02/2023</td>
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<tr>
<td>SAAB AB</td>
<td>STC</td>
<td>10078618</td>
<td>1</td>
<td>A108LUH, A109N, A109C, A109E, A109K2, A109A, A109AI</td>
<td>Inflation System for A109 Emergency Floatation System - Certification of a new inflation system as an alternative to the existing Moog Inflation System.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 27-7 initial issue. The requirements for the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this change/repair it must be determined that the sametdown would introduce no adverse effect upon the airworthiness of the product.</td>
<td>TDM21-04-MDL, issue A, dated 1 January 2023; Instructions for Continued Airworthiness, TDM21-04-4-ICAIssue A, dated 10 February 2022; Or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>25/10/2023</td>
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<tr>
<td>ALM ENGINEERING GmbH</td>
<td>STC</td>
<td>10078637</td>
<td>1</td>
<td>BD-700-1411, BD-700-1A10</td>
<td>Install of Collins AID &amp; TIM for EFB Installation of Collins single Aircraft Interface Device (AID) with dual Tablet Interface Modules (TIM) providing aircraft data in peer-to-peer communication and electrical power to Portable Electronic Flight Bag (EFB).</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.29, 25.301.25.303, 25.305, 25.307, 25.561.25.581, 25.601, 25.603, 25.605, 25.607, 25.609, 25.611, 25.613.</td>
<td>None</td>
<td>17/02/2022</td>
<td>Active</td>
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<tr>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>STC</td>
<td>10078639</td>
<td>1</td>
<td>737-900ER, 737-900, 737-9, 737-8200, 737-800, 737-8, 737-700, 737-600</td>
<td>Auxiliary Battery System - This Change alters the configuration from a single battery configuration to a dual battery configuration to increase the standby power time from 30 minutes to a minimum of 60 minutes standby power. Revision 1: Additional Models 737-800, 737-900ER, 737-8, 737-8200, 737-9.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 17.25.1.29; 30.1;30.3.30.5;30.6.30.61.5.581, 601;603;609;611;786;831;853, 602;609;611;621, 625;629, 25.150;25.25; 1:25...</td>
<td>Prior to installation of this change/repair it must be determined that the sametdown would introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Document List MDL1586-001 Issue 7, Aircraft Flight Manual Supplement FMS 1586-001 Issue 1, FMS 1586-002 Issue 1, FMS 1586-004 Issue 1, FMS 1586-008 Issue 1, FMS 1586-009 Issue 1, FMS 1586-010 Issue 1, FMS 1586-011 Issue 1, FMS 1586-012 Issue 1, FMS 1586-013 Issue 1, FMS ...</td>
<td>16/06/2022</td>
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<tr>
<td>TELAIR US LLC</td>
<td>FAA STC</td>
<td>10078646</td>
<td>1</td>
<td>ATR 72-212A A</td>
<td>Installation of a Cargo Loading System Field Section, P/N 72272-102</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.25.23, 25.321, 25.601, 25.607, 25.609, 25.619, 25.621, 25.623, 25.625, 25.150;25.25...</td>
<td>1. U.S. Cargo Systems Master Drawing List (MDL) for ATR72-600F Fed Ex System Load System: RPT-3068, Issue 4.0, dated 17 December 2021, or later FAA approved revision. 2. Illustrated Parts Catalog &amp; Component Maintenance Manual, U.S. Cargo Systems Publication No. 50-20-46, Rev. 2.0, dated 10 December 2021.</td>
<td>Prior to installation of this change/repair it must be determined that the sametdown would introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/03/2022</td>
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<tr>
<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>DHC-6 SERIES 400</td>
<td>STC-AT-151 DHC-6 IR Suppression System</td>
<td>The Type and OSD Certification Bases (CB) for the original equipment manufacturer applicable to this STC/ approval. The requirements for environmental protection and the associated/ certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certification/ approval w...</td>
<td>Master Data list Change Dossier STC-AT-151. Document Number STC-AT-151. Rev. 6( ) Aircraft Flight Manual Supplement for DHC-6 Series 400. IRSInstallation, Document Number AFM -STC...</td>
<td>15/02/2022</td>
<td>Active</td>
<td>10078650</td>
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<tr>
<td>AGUSTAWELESA ND. PHILADELPHIA, N.C.</td>
<td>AIX119 MkIV</td>
<td>Installation of crash resistant fuel tanks and other fuel system modifications. This fuel system modification includes also the installation of an additional Closed Circuit Refueling Receiver (CCRR), to allow hotfueling during ground operations with the engine running and not running.</td>
<td>The Certification Basis for the original product as amended the following additional or alternative environmental requirements: Special Condition (s): E-1 for the optional closed-circuit refueling receiver (CCRR) The following paragraph (s) at a later amendment: CS27.952(a)(1)(2)(3)(5)(6)(c)(3)</td>
<td>- Master data list: AWP- ENG-19-064 Rev A dated April 6, 2021 - Instructions for Continued Airworthiness. AWP- ENG-20-094 Rev - dated March 8, 2021. Or later revisions of the above listed document(s) approved/ accepted condition of EASA in accordance with the Technical Implementation/Procedures of E.U.</td>
<td>22/02/2022</td>
<td>Active</td>
<td>10078663</td>
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<tr>
<td>MT-PROPELLER ENTWINCKLING GmbH</td>
<td>R172i, R172j, R172k</td>
<td>Installation of MTV-12-D(180-17) I( ) or MTV-12-D(188-56) I( ) Propeller on Cessna Series and installation of MTV-12-D(188-56) I( ) Propeller on Cessna Series.</td>
<td>- Master data list: AWPC- ENG-19-064 Rev A dated April 6, 2021 - Instructions for Continued Airworthiness. AWP- ENG-20-094 Rev - dated March 8, 2021. Or later revisions of the above listed document(s) approved/accepted condition of EASA in accordance with the Technical Implementation/Procedures of E.U...</td>
<td>- According to limitation section of AFM-S. Prior to installation of this change/repair it must be determined whether the installation of this change/repair introduces any adverse effect upon the airworthiness of the product...</td>
<td>02/03/2023</td>
<td>Active</td>
<td>10078671</td>
<td></td>
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<tr>
<td>MT-PROPELLER ENTWINCKLING GmbH</td>
<td>U206E, U206F, U206G, U206A, U206B, U206C, U206D, TU206E, TU206F, TU206G, U206, TU206B, TU206C, TU206D, TP206E, TP206F, TP206A, TP206B, TP206C, P206C, P206D, P206E, T206H, 206, 206H, 206B, P206A, P206B</td>
<td>Installation of MTV-5-1-E-C-F-R(P) I( ) or MTV-206I-54( ) Propeller on Cessna 206 Series with PT6A engine conversion.</td>
<td>The Type and OSD Certification Bases for the original product as amended the following additional or alternative environmental requirements: The following paragraph(s) at a later amendment: CS38.864 Rev 6. FAR 36Amdt 36-31. This certification/approval involves a change with an impact on the noise database...</td>
<td>According to MT-Propeller Master Documentation List No. E-3752 Rev. 2( ) or later/maintained - i.e. WT-Propeller ICA Doc. No. E-3534 Rev. 2( ) applicable to this installation of this change/repair it must be determined whether the installation of this change/repair introduces any adverse effect upon the airworthiness of the product...</td>
<td>02/04/2022</td>
<td>Active</td>
<td>10078672</td>
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<tr>
<td>LUFTHANSA TECHNIC AG</td>
<td>A330-343</td>
<td>E/C INSTALLATIONPassenger cabin modification, including installation of Premium Economy Class seats.</td>
<td>The Type and OSD Certification Bases for the original product as amended the following additional or alternative environmental requirements: The following paragraph(s) at a later amendment: CS26.7V(9)(b) am10.0,CS25.853 (a)(d)(v) am10.23 The requirements for environmental protection and the associated...</td>
<td>Supplemental Type Design Document STD25-21/18M Issue 01/00Instructions for Continued Airworthiness A-06-22/08 FAOA Issue 02/00 - later revisions of the above listed document(s) approved/ accepted condition of the EASA system...</td>
<td>24/02/2022</td>
<td>Active</td>
<td>10078673</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>ST02759LA</td>
<td>F/A-18A/B/C/D</td>
<td>TME 08</td>
<td>Installation of Premium Economy Class seats between door 3 and 4 of Boeing 777-300ER Revision 1: new front row seat with an improved audio/video system/VIP Revision 2: technical documentation list corrected</td>
<td>The Type and OSD Certification Bases for the original product are amended by the following additional or alternative airworthiness requirements: The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Supplemental Type Design Document STDD-19-103M4 Issue 0200/Instructions for Continued Airworthiness A-00-211/IC A Issue 00/1000/Instructions for Continued Airworthiness A-00-220/1010/Information HLT A-00-2012-C01 Rev. 01</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/02/2022</td>
<td>Active</td>
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<tr>
<td>AVMAX AVIATION SERVICES Inc</td>
<td>DHC-8-102</td>
<td>C-LSA08-113/ D ISSUE 1</td>
<td>BSH 050</td>
<td>Retrofit Installation – Separation of Warning &amp; Caution Annunciators</td>
<td>The Type and OSD Certification Bases (CB) for the original production products applicable to this certificate/approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/02/2022</td>
<td>Active</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>777-300ER</td>
<td>ST02759LA</td>
<td>Installation of Premium Economy Class seats between door 3 and 4 of Boeing 777-300ER Revision 1: new front row seat with an improved audio/video system/VIP Revision 2: technical documentation list corrected</td>
<td>The Type and OSD Certification Bases (CB) for the original product are amended by the following additional or alternative airworthiness requirements: The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Supplemental Type Design Document STDD-19-103M4 Issue 0200/Instructions for Continued Airworthiness A-00-211/IC A Issue 00/1000/Instructions for Continued Airworthiness A-00-220/1010/Information HLT A-00-2012-C01 Rev. 01</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/06/2022</td>
<td>Active</td>
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<tr>
<td>HONEYWELL INTERNATIONAL Inc</td>
<td>REV. 1</td>
<td>777-300ER</td>
<td>ST02759LA</td>
<td>Installation of the Honeywell RDR-4000 Weather Radar System</td>
<td>The Type and OSD Certification Bases (CB) for the original production products applicable to this certificate/approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td></td>
<td></td>
<td></td>
<td>01/03/2022</td>
</tr>
<tr>
<td>INTERCONTINENTAL TAL SERVICECORPORATION</td>
<td>REV. 1</td>
<td>FAA SAO16282</td>
<td>ANTI II</td>
<td>Installation of ADS-B Out Rev. 1 corrects typographical errors in AFM reference.</td>
<td>The Type and OSD Certification Bases (CB) for the original production products applicable to this certificate/approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>1. Avionics Integration Master Data List, No. 31-318L001, Rev. D dated 10/ May 2015. Avionics Integration Aircraft Flight Manual Supplement, No. 31-348B001, Rev. IR dated 16 May 2015, or later revisions to 1 or 2 above listed document approved accepted on behalf of EASA in accordance with the Technical I...</td>
<td>Limited to S/N 1105 and On equipped with Rockwell Collins Proline 21/Avionics. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the...</td>
<td>22/03/2022</td>
<td>Active</td>
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<td>EASA STC #</td>
<td>STC Holder</td>
<td>STC Ref.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>10078695</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING SERVICES INC.</td>
<td>STC No. 1</td>
<td>10078695</td>
<td>767-300</td>
<td>B767-300 Cabin Layout Modification/Sealing Layout Change to increase the Premium Economy Class (E) seat capacity in a 3 class seating configuration.</td>
<td>The Type and OSD Certification Bases (CB) for the original product/manufacturer and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document AVI-02500-MDL-001</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair is compatible with any other previously installed change and/or repair.</td>
<td>02/03/2022</td>
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<tr>
<td>10078696</td>
<td>PI-NAV AVIATION SERVICES INC.</td>
<td>STC Ref. 1</td>
<td>10078696</td>
<td>320-321/-322/-323/-341/-342</td>
<td>ELT Survival Installation</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The existing Weather Radar receiver/transmitter and antenna system.</td>
<td>Master Document AVI-02500-MDL-001</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair is compatible with any other previously installed change and/or repair.</td>
<td>02/03/2022</td>
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<tr>
<td>10078697</td>
<td>AVMAX AVIATION SERVICES INC.</td>
<td>22_MTH_04082021</td>
<td>10078697</td>
<td>SA14-35, ISSUE NO.: 1</td>
<td>Weather Radar Upgrade</td>
<td>This modification replaces an existing Weather Radar system with a Rockwell Collins TWR-850 System.</td>
<td>Master Document AVI-02500-MDL-001</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair is compatible with any other previously installed change and/or repair.</td>
<td>02/03/2022</td>
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<td>10078700</td>
<td>AVIATION TRADERS LIMITED EUROPE S.A.S.</td>
<td>22_MTH_04082021</td>
<td>10078700</td>
<td>737-800</td>
<td>BOEING 737-800 INSTALLATION OF MADELEC AERO LED CABIN LIGHTING</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): CRI H-01 EWIS ICA</td>
<td>Master Document AVI-02500-MDL-001</td>
<td>None</td>
<td>03/03/2022</td>
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<td>10078701</td>
<td>PILATUS BUSINESS AIRCRAFT, LTD.</td>
<td>22_MTH_04082021</td>
<td>10078701</td>
<td>FAA STC SA01082DE</td>
<td>Installation of Cabin Power Outlets</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Document AVI-02500-MDL-001</td>
<td>This STC is applicable to all serial numbers for the Pilatus PC-24 aircraft. Prior to installation of this change/repair it must be determined that the change/repair is compatible with any other previously installed change and/or repair.</td>
<td>03/03/2022</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAATSTC ref.</td>
<td>Movie</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10078702</td>
<td>ROCKWELL</td>
<td></td>
<td>ST02691SE</td>
<td>BD-100-1A10</td>
<td>Installation of Head-up Display System (HUD) Model 6500 and EnhanceVision System (EVS) Model 3560 on the Bombardier Model BD-100-1A10 (Challenger 350) aircraft.</td>
<td>The Type and OSD Certification Bases (CB) for the original production chemistries applicable to this certificate/appraisal. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/appraisal.</td>
<td>FAA STO ST02691SE</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/02/2022</td>
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<td>10078713</td>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td></td>
<td>TCCA-10517-122</td>
<td>B300C, B200C, BG-100FG-1</td>
<td>Installation of the Medivac Audio System as per TCCA STC TCCA-10517-122</td>
<td>The Type and OSD Certification Bases (CB) for the original production chemistries applicable to this certificate/appraisal. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/appraisal.</td>
<td>FAA STO</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/03/2022</td>
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<tr>
<td>10078716</td>
<td>AEROSERVIS, S.R.O.</td>
<td></td>
<td>L-410 UVP-E, L-410 UVP-E/L, L-410 UVP-E20, L-410 UVP-E/20, L-410 UVP-E20 CARGO, L-420</td>
<td>STC AE219 [L-410 MLA rear hinge]</td>
<td>The Type and OSD Certification Bases (CB) for the original production chemistries applicable to this certificate/appraisal. Applicable to following models: L-410 UVP-E, L-410 UVP-E20, L-410 UVP-E/20, L-410 UVP-E20 CARGO, L-420.</td>
<td>The Type and OSD Certification Bases (CB) for the original production chemistries applicable to this certificate/appraisal. Applicable to following models: L-410 UVP-E, L-410 UVP-E20, L-410 UVP-E/20, L-410 UVP-E20 CARGO, L-420.</td>
<td>FAA STO</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/03/2022</td>
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<tr>
<td>10078748</td>
<td>WEST STAR AVIATION, L.L.C.</td>
<td></td>
<td>FAA STC ST04519CH</td>
<td>750</td>
<td>Installation of a Honeywell Flight Management System Software Update to version NZ 6.2.1</td>
<td>The Certification Basis for the original product remains applicable to this certificate/appraisal. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged/remain applicable to this certificate/appraisal.</td>
<td>FAA STO</td>
<td>This STC is applicable to the following serial numbers: MSN750-000 through 750-0500. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/03/2022</td>
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<tr>
<td>10078750</td>
<td>SABENA TECHNICS BOC</td>
<td></td>
<td>777F</td>
<td>777-000ER, 777-300, 777-300L, 777-200</td>
<td>Mounting device and power supply connection for ‘PIVOT’ EFB/solutions.</td>
<td>The Type and OSD Certification Bases (CB) for the original production chemistries applicable to this certificate/appraisal. The following additional or alternative airworthiness requirements:</td>
<td>Master Documents List 10004-01-A-MDL, Rev 0; AFM Supplement 10004-01-A-AFM5-R02, Rev 2; Instructions for Continued Airworthiness 10004-01-A-ICA Rev 0; MEL Supplement 10004-01-A-MLERev 0</td>
<td>As per AFM supplement. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/03/2022</td>
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<td>STC Holder</td>
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<td>Description</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>STC Holder</td>
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<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>UUIDS AERO CONCEPT</td>
<td>REV. 1</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>UNIVERSAL CONVERTIBLE STRETCHER INSTALLATION</td>
<td>Extension of STC 10078751 to an AML STC for installation of medical stretcher on-board</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Per AFM supplement as identified in the EASA APPROVED MODEL LIST. Installation is limited to configurations as per below Installation instructions and Limitations Drawings: UAC21M001-101, UAC21M001-M0002, UAC21M001-M0003, UAC21M001-M0004, UAC21M001-M0005, UAC21M001-M0006, UAC21M001-M0007, UAC21M001-M0008.</td>
<td>04/02/2023</td>
<td>Active</td>
<td>10078751</td>
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<tr>
<td>ETHAD AIRWAYS ENGINEERING L.L.C.</td>
<td></td>
<td>A321-272N,-271N</td>
<td>CARRIAGE OF CARGO IN PASSENGER CABIN</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: [\text{EASA validation of cockpit door bubble searchlight on a side mount or center of}} ]</td>
<td>Master Document List</td>
<td>EA2-20251298-DL01 Issue 3AFM Supplement EA2-20251298-P048 Issue SAN-20251298-P048 Issue 10SD CDQ Supplement EA2-20251298-TR03 Issue 30SD MML Supplement EA2-20251298-P11 Issue 4or later revisions of the above listed document(s) apply.</td>
<td>- The allowance to transport cargo in passenger compartments limited to 2000 kg following installation of the change or to July 31st 2022 whatever occurs first. - Airplanes modified per this STC are considered to be a &quot;variant&quot; to Type 777-300ER passenger configurations. Cabin crew operating the aircraft.</td>
<td>08/03/2022</td>
<td>Active</td>
</tr>
<tr>
<td>VIH AEROSPACE Inc.</td>
<td>REV. 1</td>
<td>AS 332 L, AS 332 L1</td>
<td>EASA validation of cockpit door bubble searchlight installationRev. 1: Addition of AS332 L2.</td>
<td>The Type and OSD Certification Bases for the original product as determined by the following additional or alternative airworthiness requirements:</td>
<td>Project Completion Plan, rev. VNH 2301-001, Rev A, dated 14/09/2022; rotorcraft Flight Manual (PFM) Supplement, ref. V332-1715-FMS, Rev C, dated 29/03/2022; Instructions for Continued Airworthiness (ICA), ref. V332-1715-ICA, Rev B, dated 19/05/2022; Master Document List (MDL), ref. V332-1715-MDL, Rev B, date...</td>
<td>Prior to installation of this change/repair it must be determined whether the time limit for factory modification is exceeded or the aircraft is first flown on the date of initial operation or later revisions of the above listed document(s) apply.</td>
<td>08/12/2022</td>
<td>Active</td>
<td>10078795</td>
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<tr>
<td>AVIONIK STRAUBING ENTWICKLINGS GmbH</td>
<td></td>
<td>P210R</td>
<td>Installation of dual Garmin GA 58 TAS/TCAS antennas</td>
<td>The Type and OSD Certification Bases for the original product as determined by the following additional or alternative airworthiness requirements: [\text{installation in accordance with Avionik Straubing GmbH Instructions for Installation. Doc No: ASR-2021-050-PI-10-00, date 01/02/22. Instructions for Continued Airworthiness in accordance with Avionik Straubing GmbH, Doc No: ASR-2021-060-ICA-11-00, date 31/01/22 or later revisions of the above listed do...}</td>
<td>Installation in accordance with Avionik Straubing GmbH Instructions for Installation. Doc No: ASR-2021-050-PI-10-00, date 01/02/22. Instructions for Continued Airworthiness in accordance with Avionik Straubing GmbH, Doc No: ASR-2021-060-ICA-11-00, date 31/01/22 or later revisions of the above listed do...</td>
<td>Prior to installation of this change/repair it must be determined whether the time limit for factory modification is exceeded or the aircraft is first flown on the date of initial operation or later revisions of the above listed document(s) apply.</td>
<td>10/03/2022</td>
<td>Active</td>
<td>10078797</td>
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<tr>
<td>AGUSTA WESTLAND PHILADELPHIA AIRCRAFT CORPORATION</td>
<td></td>
<td>A911/9MKII</td>
<td>Installation of a Spectrolab SX-5 searchlight on a side mount or center of nose</td>
<td>The Certification Basis for the original product remains applicable with no additional or alternative airworthiness requirements:</td>
<td>Agusta Aerospace Corporation Master Data List, Report No: ACC-ENG-00-036, Rev. 7, dated 28 April 2011 for configuration 1 (A119/Only), Agusta Aerospace Corporation Instructions for Continued Airworthiness Report: No. AAC-ENG-00-036, Rev. 7, dated 2 March 2001 for configuration 1(A119/ Only). Agusta Aeros...</td>
<td>Prior to installation of this change/repair it must be determined whether the time limit for factory modification is exceeded or the aircraft is first flown on the date of initial operation or later revisions of the above listed document(s) apply.</td>
<td>29/03/2022</td>
<td>Active</td>
<td>10078802</td>
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**EASA STC list:**

- **STC Holder:** UUIDS AERO CONCEPT
- **Model:** SEE EASA APPROVED MODEL LIST
- **Description:** UNIVERSAL CONVERTIBLE STRETCHER INSTALLATION
- **Certification Basis:** Extension of STC 10078751 to an AML STC for installation of medical stretcher on-board
- **Limitations/Conditions:** Per AFM supplement as identified in the EASA APPROVED MODEL LIST. Installation is limited to configurations as per below Installation instructions and Limitations Drawings: UAC21M001-101, UAC21M001-M0002, UAC21M001-M0003, UAC21M001-M0004, UAC21M001-M0005, UAC21M001-M0006, UAC21M001-M0007, UAC21M001-M0008.
- **Date of Issue:** 04/02/2023
- **Status:** Active
- **System Record:** 10078751

- **STC Holder:** ETHAD AIRWAYS ENGINEERING L.L.C.
- **Model:** A321-272N,-271N
- **Description:** CARRIAGE OF CARGO IN PASSENGER CABIN
- **Certification Basis:** The Type and OSD Certification Bases for the original product as determined by the following additional or alternative airworthiness requirements: Deviation(s): CRI 0-130 "Transportation of cargo in passenger compartments" Issue 3 The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.
- **Limitations/Conditions:** Prior to installation of this change/repair it must be determined whether the time limit for factory modification is exceeded or the aircraft is first flown on the date of initial operation or later revisions of the above listed document(s) apply.
- **Date of Issue:** 08/03/2022
- **Status:** Active
- **System Record:** 10078753

- **STC Holder:** VIH AEROSPACE Inc.
- **Model:** AS 332 L, AS 332 L1
- **Description:** EASA validation of cockpit door bubble searchlight installationRev. 1: Addition of AS332 L2.
- **Certification Basis:** The Type and OSD Certification Bases for the original product as determined by the following additional or alternative airworthiness requirements: Deviation(s): CRI 0-130 "Transportation of cargo in passenger compartments" Issue 3 The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.
- **Limitations/Conditions:** Prior to installation of this change/repair it must be determined whether the time limit for factory modification is exceeded or the aircraft is first flown on the date of initial operation or later revisions of the above listed document(s) apply.
- **Date of Issue:** 08/12/2022
- **Status:** Active
- **System Record:** 10078795

- **STC Holder:** AVIONIK STRAUBING ENTWICKLINGS GmbH
- **Model:** P210R
- **Description:** Installation of dual Garmin GA 58 TAS/TCAS antennas
- **Certification Basis:** The Type and OSD Certification Bases for the original product as determined by the following additional or alternative airworthiness requirements: Deviation(s): CRI 0-130 "Transportation of cargo in passenger compartments" Issue 3 The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.
- **Limitations/Conditions:** Prior to installation of this change/repair it must be determined whether the time limit for factory modification is exceeded or the aircraft is first flown on the date of initial operation or later revisions of the above listed document(s) apply.
- **Date of Issue:** 10/03/2022
- **Status:** Active
- **System Record:** 10078797

- **STC Holder:** AGUSTA WESTLAND PHILADELPHIA AIRCRAFT CORPORATION
- **Model:** A911/9MKII
- **Description:** Installation of a Spectrolab SX-5 searchlight on a side mount or center of nose
- **Certification Basis:** The Certification Basis for the original product remains applicable with no additional or alternative airworthiness requirements: Deviation(s): CRI 0-130 "Transportation of cargo in passenger compartments" Issue 3 The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.
- **Limitations/Conditions:** Prior to installation of this change/repair it must be determined whether the time limit for factory modification is exceeded or the aircraft is first flown on the date of initial operation or later revisions of the above listed document(s) apply.
- **Date of Issue:** 29/03/2022
- **Status:** Active
- **System Record:** 10078802
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<th>Model</th>
<th>Description</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10078809</td>
<td>ALAMO ENGINEERING GmbH</td>
<td>DORNIER 328-100</td>
<td>Certify Dornier 328-100 as PED tolerant</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The paragraph (b) at a later amendment: CS 25.1319, Amend. 7. The requirements for environmental protection and the associated certificated noise and/ or emissions...</td>
<td>Master Document List, MDL-21A47AE-03, Rev. A Aircraft Flight Manual Supplements, FMS21A47AE-01, Rev. A Aircraft Flight Manual Supplements, KA21A47AE-03, Rev. A or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>20/06/2022</td>
<td>Active</td>
<td>10078823</td>
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<tr>
<td>10078814</td>
<td>EIRETCH AVIATION SERVICES LIMITED</td>
<td>737-600, 737-700, 737-800</td>
<td>Certify Dornier 328-100 as PED tolerant</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: JAR 25 Change 14.25.1(c)(7): 601.663.853.1301.1320.1326.1331.1332.1335.1336.1337.1338.1340.1419.1431...</td>
<td>Master Document List, MDL-1585-001 Issue 6r ((\approx) 6r) or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>11/03/2022</td>
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<td>10078814</td>
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<td>10078816</td>
<td>SKANDIA, INC</td>
<td>FAA STC, ST04556CH</td>
<td>Installation of T/E Audio kits</td>
<td>The Certification Basis for the original product remains applicable as defined in the certification basis with the following additional or alternative airworthiness requirements: Installation of T/E Audio kits ...</td>
<td>Skanda Master Data List, Document Number 7803-068650, Revision IR dated August 1, 2021 or later FAA approved revisions.</td>
<td>14/03/2022</td>
<td>Active</td>
<td>10078816</td>
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<tr>
<td>10078823</td>
<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>FAA STC, ST041001-042</td>
<td>Installation of a Back-up Communications System</td>
<td>The Certification Basis for the original product remains applicable as defined in the certification basis with the following additional or alternative airworthiness requirements: Installation of a Back-up Communications System...</td>
<td>Chippewa Aerospace Master Drawing List (MDL) DB-2371-0001 Rev. IR, dated 12 March or later FAA approved revision.</td>
<td>14/03/2022</td>
<td>Active</td>
<td>10078823</td>
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<td>10078824</td>
<td>U.S. TECHNICAL CONSULTANTS, INC</td>
<td>FAA STC, ST04257CH</td>
<td>Cabin installations: installation of a 16 passengers configuration installation of additional or applicable items as determined to be determined to be the only configuration in the following configuration: 22 passengers configuration...</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25.1319, Amendment 7. The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product as amended remain and remain applicable to this certificate/ approval.</td>
<td>16 Passengers configuration only - Technical Master Documents and Drawing List (MDL), 9486-25-000, Revision E, dated 17 November 2021 - Aircraft Flight Manual Supplement, U10-0003, Revision A. dated 19 July 2021 - Instructions for Continued Airworthiness, U10-0003, Revision NC, dated 13 July 2022...</td>
<td>01/06/2023</td>
<td>Active</td>
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<td>GULFSTREAM AEROSPACE CORP.</td>
<td>Installation of GoGo Business Aviation Avance LS System on the Gulfstream G-V and G-VIII aircraft (FAA STC ST04307AT-D)</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition(s): F-110 Installation of non-rechargeable Lithium Battery. The requirements for environmental protection and the associated identified noise...</td>
<td>1. Index List CE4130289018 Rev. C, dated Mar 18 20163. Instructions for Continued Airworthiness: - GIV-X Gogo Biz AVANCE LS System Installation - CE4130289020 Rev. C, dated Feb 22 20183. Instructions for Continued Airworthiness: - GIV-G V-G Biz AVANCE LS System Installation</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/03/2022</td>
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<td>GULFSTREAM AEROSPACE CORP.</td>
<td>Installation of Custom Interior on GIV-X aircraft (FAA STC ST03118AT-D)</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition(s): F-110 Installation of non-rechargeable Lithium Battery. The requirements for environmental protection and the associated identified noise...</td>
<td>Index List GC4280088648 Rev. C, dated Feb 18 2022 for Continued Airworthiness GC4280104014 Rev. C, dated Aug 19 2009 and Instructions for Continued Airworthiness CE428064027 Rev. C, dated Feb 17 2022 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The following STCs are a required part of this Custom Interior STC: ST02948AT-D Cabin Window Shades (EASA.IM.A.S.01687) ST02951AT-D Passenger Oxygen System (EASA.IM.A.S.01689) ST02957AT-D Interior Emergency Lights (EASA.IM.A.S.01684). Prior to installation of this change/repair it must be determined that...</td>
<td>17/03/2022</td>
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<td>JET AVIATION AG</td>
<td>B777-200LR provisions installation for a Dual Iridium Antenna on the B777 fuselage upper lobe.</td>
<td>The Type and OSD Certification Bases (C8) for the original productprecludes applicability to this certificate/approval. The requirements for environmental protection and the associated identified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>JB/SN-M-01277DOC-001 Issue A – Declaration of Compliance: JBSN-M-01277A001 Issue A – Certification Compliance Sheet for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/03/2022</td>
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<td>10078866</td>
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<td>SABENA TECHNICS BGC</td>
<td>Cabin Reconfiguration to 353 YC Pax</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition(s): F-110 Installation of non-rechargeable Lithium Battery. The requirements for environmental protection and the associated identified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>- Master Data List, Ref. 40930-01-A-MLD-R00 (18 Mar 2022) Instructions for continued airworthiness: 40930-01-A-ICA-R00 (19Nov 2021) for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>As a concurrent requirement of this change, accomplishment of this STC requires the simultaneous accomplishment of the following modification: Installation of a photo-luminescent EFPM as per Lufthansa Technik EASA STC No. 101394 Rev.03. Prior to installation of this change/repair it must be determined that...</td>
<td>21/03/2022</td>
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<td>INTER-TEC AERO DESIGN &amp; CERTIFICATION LTD</td>
<td>Introduction of Mid-Continent Standby Altitude Module (SAAM DM302 Series) in accordance with Master Data List IT S-0594-007</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition(s): F-110 Installation of non-rechargeable Lithium Battery. The requirements for environmental protection and the associated identified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>- Master Data List, Ref. IT S-0594-007/001 Issue 1 - Rotorcraft Flight Manual Supplement IT S-0594-006 / 001 Issue 1 - Master Minimum Equipment List Supplement IT S-0594-016 Issue 1 - Instructions for Continued Airworthiness IT S-0594-005 Issue 1 for later revisions of the above listed document(s) approved/accepted under the...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/03/2022</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Date of Issue</th>
<th>Status</th>
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<tr>
<td>AVIADECO GmbH</td>
<td>Removal of the Ransbach Air Fuzelage Stowage Locker, previously installed by the FAA STC ST00179SE</td>
<td>Removal installation in accordance with Engineering Order EO-02421-1 rev00 and EO-02421-2 rev 00; maintained in accordance with ICAO-02421 rev 00 (or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<td>GVH AEROSPACE (IRELAND) LIMITED</td>
<td>SATCOM Installation with Skytrac ISAT-2004 flight following/ satCom/Communication system and D-AL-200 Dynamic Air link (wireless network/device)</td>
<td>Certification Programme GVH-11987-01 CFM-01 Issue 3</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/03/2022</td>
<td>Active</td>
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<td>FIELD AVIATION EAST, LTD.</td>
<td>Honeywell Primus 660 Weather Radar System Replacement of existing weather radar system and display with a newer honeywell Primus 660 and Universal MOD-640 Multifunction Display</td>
<td>For Model DHC-8-102 Installation shall be performed in accordance with Field Aviation Company Inc. Master Drawing and Data list designated 25600000, install/revision/Maintenance shall be performed in accordance with Field Aviation Company Inc. Maintenance Manual Supplement designated 2560200, install...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/03/2022</td>
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<td>HELI-ONE (NORWAY) AS</td>
<td>Nose mounted FLIR installation (ref. modification AHSE-72102)</td>
<td>Certification Programme (CP) ref. AHSE-72105 Revision F dated 21/03/2022 Flight Manual Supplement (FMS) for nose mounted FLIR installation ref H1No.1.332 Revision dated 21/03/2022 Product Change Declaration (PDC) ref. AHSE-72102 Revision B dated 21/03/2022 Instructions for Continued Airworthiness (IC) for...</td>
<td>The following modifications are a prerequisite for the embodiment of this STC - Heli-One (Norway) AS ref. AHSE-73030 and AHSE-73045 - Mission panel modifications; Aerosol Helicopters ref. 875624 - Provisions for installation of FLR system Laser pointer (laser class 3B) and laser illuminator (laser c...)</td>
<td>23/03/2022</td>
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<td>WIPARE Inc.</td>
<td>Installation of Wipaire Model 6100 Amphibious/Seaplane Floats</td>
<td></td>
<td>This float installation is compatible with aircraft incorporating Pilatus underwing fuel tanks, part numbers 950·17·11·103 (2 x 40 U_Sgallons), 115.55.06.225 (2 x 83 US gallons), or 115.55.06.305 (2 x 64 US gallons). The fuel tank installations are approved on Type Certificate Data Sheet TA15. Prior to...</td>
<td>24/03/2022</td>
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<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
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<td>AVIONIK ENTWICKLUNG GmbH</td>
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<td>Garmin GTN 60X (X)</td>
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<td>SA26-AT</td>
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<td>Garmin GTN 7XX Installation</td>
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<td>SA26-AC</td>
<td>Garmin GA 35 GPS antenna</td>
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<td>Garmin GA 35 GPS antenna installation</td>
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<td>JET AVIATION AG</td>
<td>BD-700-1A10</td>
<td>Cabin LOPA change</td>
<td>Accredited</td>
<td>Garmin GTN 60X Installation</td>
<td>23.571(c)</td>
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<td>COMTECH GLOBAL AVIATION SERVICES, SOCIE DAD LIMITADA</td>
<td>10078948</td>
<td>757-200, 757-200PF, 757-300</td>
<td></td>
<td>Installation of Reticulated Polyurethane Foam (RPF) for Center Fuel Tank Ignition Mitigation in the Boeing 757-200 &amp; 757-300 Series Aircraft</td>
<td>This STC installs a Fully Packaged Reticulated Polyurethane Foam (RPF) avionics Fuel Tank Ignition Mitigation Means (MM) in the Center Fuel Tank (CF757-200-300)</td>
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<tr>
<td>ALAMO ENGINEERING GmbH</td>
<td>10078957</td>
<td>BD-700-1A11</td>
<td></td>
<td>Installation of Ku-Band Satcom with TMA that installs covers the upgrade of the existing satellite/communications system by a Satcom Direct Plane Simple Ku-Band Satcom System consisting of: a Satcom Direct Modem Unit (SMU) in the aft equipment bay and an Advanced Tail Mounted Antenna (ATMA) on existing...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.13.19 at amd0020. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>AER SALE INC</td>
<td>10078962</td>
<td>FAA ST04203NY</td>
<td></td>
<td>Installation of Reticulated Polyurethane Foam (RPF) for Center Fuel Tank Ignition Mitigation</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.3060, 25.3061(a)(6), 25.3061(b)(1)(ii), 757-200, 757-200PF, 757-300.</td>
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<td>FIELD AVIATION COMPANY INC</td>
<td>10078974</td>
<td>C-LSA03-146/ D</td>
<td></td>
<td>Installation of Reticulated Polyurethane Foam (RPF) for Center Fuel Tank Ignition Mitigation</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.3060, 25.3061(a)(6), 25.3061(b)(1)(ii), 757-200, 757-200PF, 757-300.</td>
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<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>10079001</td>
<td>BD-700-1A10</td>
<td></td>
<td>Installation of a new cabin configuration and associated equipment.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.3060, 25.3061(a)(6), 25.3061(b)(1)(ii), 757-200, 757-200PF, 757-300.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>SIST FLUGTECHNIK GmbH</td>
<td>External Intelligence, Surveillance and Reconnaissance (ISR) Provisions</td>
<td>The Type and OSD Certification</td>
<td>Installation according to document “STC Holder Specific Data” with minor changes</td>
<td>Flight into known icing conditions is prohibited. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/04/2022</td>
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<tr>
<td>LEE AEROSPACE, INC.</td>
<td>Installation of LAL Windsiedel Panel FRev. 1 corrects the applicability.</td>
<td>The Type and OSD Certification</td>
<td>Master Data List IAI-MDL-119, Revision III, dated 02 September 2021 of Instructions of</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/04/2022</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>A321-Cabin Reconfiguration from 168PAX to 220PAX</td>
<td>The Type and OSD Certification</td>
<td>Master Data List IAI-MDL-119, Revision III, dated 02 September 2021 of Instructions of</td>
<td>As a concurrent requirement of this change, accomplishment of the following modification: Installation of a photo-luminescent EPM as per Luftfahrttechnik Engineering Bulletin ref A320-EB33-0384 approved through EASA STC No. 10014013. Prior to install...</td>
<td>15/07/2022</td>
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<tr>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>Passenger Salas Raarrangement/Craft with initial LOPA (377) changed due to new seats part numbers. Seat locations remain unchanged from the previous version. LOPA, only the part numbers assigned. New seats PROTEUS and REGIONAL AIRLINES Series comply with TSO-C22a and TSO-C172a.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraphs at a later amendment: JAR25 at paragraphs 111, 1441 and C25 at amended parts C25 at amended paragraphs: 29, 301, 303, 304, 305.</td>
<td>Master Data List IAI-MDL-119, Revision III, dated 02 September 2021 of Instructions of</td>
<td>As a concurrent requirement of this change, accomplishment of the following modification: Installation of a photo-luminescent EPM as per Luftfahrttechnik Engineering Bulletin ref A320-EB33-0384 approved through EASA STC No. 10014013. Prior to install...</td>
<td>11/04/2022</td>
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<tr>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A319-153N Interior Outfitting. The change consists of a cabin interior outfitting of an A319-153N with 21 passengers VIP interior for private use only.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements. Special Condition(s):</td>
<td>MODLOPA Modification in EMB-135 Ref. 145_520-18-02 of 31/03/2022.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/04/2022</td>
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<td>10070400</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A350-941</td>
<td></td>
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<td>DHU-A350 Cabin Interior Adaptation - Cabin modification and re-branding to Deutsche Lufthansa (DLH) standard.</td>
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<td>10079044</td>
<td>SABENA TECHNICS BOD</td>
<td>REV. 1</td>
<td>737-800</td>
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<td>CHOPIN 3 cabin modificationRev 1: Tyto corrected in the technical documentation list</td>
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<td>HONEYWELL INTERNATIONAL Inc.</td>
<td>REV. 1</td>
<td>FAA SR04172Ch</td>
<td>AW139</td>
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<td>10079050</td>
<td>AMTECH AERONAUTICAL LIMITED</td>
<td>TCCA S19-109 ISSUE 2</td>
<td>S2R-T34, S2R-T690</td>
<td>Installation of a Pratt and Whitney Engine Services, Inc. Flight Data/ Acquisition Storage and Transmission System (FAST)</td>
<td>The Type and OSD Certification Basis (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
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<td>10079062</td>
<td>AVIO DESIGN GROUP INC dba AVIO DO</td>
<td>SH21-4 ISSUE 1</td>
<td>412EPP</td>
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<td>Limitations/Conditions</td>
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<tr>
<td>HAMBURG GmbH</td>
<td>Installation of ADF Antenna with Antenna Adapter Plate</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List Document No. 617-10023-MDL-0001 Rev.7, Aircraft Maintenance Manual Supplement Document 617-10023-AMM-0001 Rev.8, Instructions for Continued Airworthiness Document 617-10023-ICA-0001Rev.8, BManufacturing Instructions Document 617-10023-MIN-0001 Rev.</td>
<td>Only applicable to Beech B300 S/N FL-630 RA-02812 Prior to installation of this change/repair it must be determined in the AIR notice of change that the interrelationship between this change/repair and any other previously installed change/repair will not introduce adverse effect upon the airworthiness of the product.</td>
<td>20/04/2022</td>
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10079077 JET AVIATION AG

SEE EASA APPROVED MODEL LIST

MYSTERE FALCON 900 (MF900)
MYSTERE FALCON 50 (MF50)
MYSTERE FALCON 20 (GF)
MYSTERE-FALCON 20 - F5
MYSTERE-FALCON 20 - E5
MYSTERE-FALCON 20 - D5
MYSTERE-FALCON 20 - C5
MYSTERE-FALCON 200 GV-SP
GVH (GE50)
GVH X GVH
GII (G-1159A)
GII (G-1159B)
GII (G-1159B1)
G-199
FAN JET FALCON SERIES G
FAN JET FALCON SERIES F
FAN JET FALCON SERIES E
FAN JET FALCON SERIES D
FAN JET FALCON SERIES C
FAN JET FALCON FALCON 900EX (F900EX)
FALCON 7X
FALCON 200EX
FALCON 2000
DHC-8-402
DHC-8-401
DHC-8-315
DHC-8-314
DHC-8-311
DHC-8-301
DHC-8-202
DHC-8-201
DHC-8-106
DHC-8-103
DHC-8-102
CL-600-2E25 (RU SERIES 1000)
CL-600-2D24 (RU SERIES 900)
CL-600-2D15 (RU SERIES 705)
CL-600-2C10 (RU 700, 701, 702)
CL-600-2B19 (RU SERIES 100)

Real Glass Window Shade Lens, this design change replaces all thermoplastic transparent protective window covers in the passenger cabin with a laminated glass product. In addition the lens replacement, structural change may be applied to the window shades. The attachments to the A/C structure will...

Jet Aviation Approved Model List ref. JBSL-M-0145SML-01 issue 03 dated 02.02.2022 Master Data List ref. JBSL-M-0149MDL-01 issue 03 dated 31.03.2022 Declaration of compliance ref. JBSL-M-0144DOC-01 issue 04 dated 01.03.2022 or later revisions of the above listed document(s) approved/ accepted under the EASA AML reported in annex to this certificate. However, the design/ configurations approved for installation are limited to those listed in the Jet Aviation Approved Model List JBSL-M-0114SML-01 Appendix A at Issue 3 (or subsequent revisions)...

This STC applies to the aircraft types and models listed in the EASA AML approved in annex to this certificate. However, the design/ configurations approved for installation are limited to those listed in the Jet Aviation Approved Model List JBSL-M-0114SML-01 Appendix A at Issue 3 (or subsequent revisions)...

21/04/2022 Active 10079077
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Real Glass Window Shade Lens
### List of EASA Supplemental Type Certificates

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<td>AMES AZOS Aerospace &amp; Mechanical Engineering Services Inc.</td>
<td>A320-216, A320-214</td>
<td>737-300</td>
<td>ANCRA International LLC</td>
<td>Cabin Modification/Seating layout change to change the installed seat models and increase the seating capacity</td>
<td>The Type and OSD Certification Bases (CB) for the original products/airspace applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>The change is defined with revision 0 of AMES Engineering Order DocumentAO_M2760101_S_RSP_02/12/07The installation shall be accomplished in accordance with revision 0 of AMES Instructions for Continuous Airworthiness DocumentICA_M2760101_S_RSP_271220/220 for later revisions of the above listed documents.</td>
<td>The modification is only applicable to the basic aircraft model without any MOD as listed on the EASA TCDS is installed/Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10079092</td>
<td>L2 Consulting Services Inc.</td>
<td>FAA STC ST0995AC</td>
<td>737-700/800/900/-900ER</td>
<td>Installation of Panasonic FlightLink</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) of a later amendment CSS5 Amend. 0, 0.29, 25.101, 25.103, 25.105, 25.119, 25.121, 25.125, 25.251, 25.301, 25.303, 25.305, 25.307, 25.365, 25.561...</td>
<td>Master Drawing List 3170-B107-MDL, Revision A, dated 2 November 2016&lt; (&gt;,&lt;) or later FAA approved revision Airplane Flight Manual Supplement, Document 3170-B107-AMF, Revision B, dated 10 January 2017, or later FAA approved revision Instructions for Continuous Airworthiness, Document 3170-B107-ICA Rev...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>Ancra International LLC</td>
<td>FAA STC ST2762LA</td>
<td>A320-233, A320-232, A320-231, A320-216, A320-214, A320-212, A320-211</td>
<td>Installation of main deck cargo loading systemRev. 1 adds two additional manufacturer serial numbers (MSN), 2164 and2288, to the Limitations and Conditions section.</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Ancra International LLC Control Data List DL0559, Revision B, dated 27/December 2021, or later FAA approved revision Airplane Flight Manual Supplement for Airbus A320, report number 17/3401, Revision IR, dated 28 January 2022, or later FAA approved revision Instructions for Continuous Airworthiness ICA...</td>
<td>The installation is for serial numbers MSN 2724, 2164 and2288 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>GVI (G650)</td>
<td>Installation of GoGo Business Avenza L5 System on Gulfstream G650</td>
<td>The Type and OSD Certification Bases (CB) for the original products/airspace applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>- FAA STC ST0437AT-D Index List, GDL 100000000, Revision D, dated 15 April 2022- EASA Validated Configurations List ST0437AT-D, GCD2000023156 Rev ...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/04/2022</td>
<td>Active</td>
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Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.

For the operation of the Neonatal transport system on the AW169 helicopter to be equipped with – KIT EMS POWER SOCKET, p/n 62646/004; and – EMS MacPherson floor, p/n 62633/2003, and – 8 Foldable Seats Internal Arrangement p/n 62625/01111 on alternately 7 Fc...
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<th>STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<td>10079104</td>
<td>AIRBUS S.A.S.</td>
<td>A320-214, A320-216</td>
<td>Installation of a 180 pax seat layout</td>
<td>The Type and OSD Certification Bases (CB) for the original product; the associated certification/approval. The requirements for environmental protection and the associated/related noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Summary Report ref: SRSA-21-044 Issue 4</td>
<td>Modification Approval Sheet ref. MAS A0229 Issue 1 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherwise previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>369 AEROSPACE OOS LIMITED</td>
<td>REV. 1</td>
<td>737-800, 737-900ER 737-700, 737-800</td>
<td>Introduction of FANS 2 Data Link Capabilities (Dual Stack)/Rev.1 corrects reference of Technical Documentation.</td>
<td>The Type and OSD Certification Bases (CB) for the original product; the associated certification/approval. The requirements for environmental protection and the associated/related noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>MDL-1502 Master Data List Revision DFMS-1502-001 Introduction of FANS 2 (Dual Stack) Revision CirCA-1502-001 Instructions for Continued Airworthiness (SA) Rev later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This approval is only applicable to aircraft that have previously been equipped with a CPDLC (ATN-811) System installed by STC 1004849 Revision2 (or later approved revision) or by the aircraft Type Certificate Holder using Collins Aerospace CMU-900 in addition, a SATCOM System installed by STC 1004903...</td>
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<td>AS 332 L1 AS 332 L AS 332 C1 AS 332 C</td>
<td>Installation of a Garmin GTN650 &amp; GTN750 NAV / COM / GPS System</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended thereby the following additional or alternative airworthiness requirements: The following paragraph(s) of a later amendment: the following paragraph(s) of a later amendment: the following paragraph(s) of a later...</td>
<td>Master Document List, Doc. No. HAU-D-056-001, Revision 1; Installation Manual, Doc. No. HAU-D-056-004, Revision 1; Flight Manual Supplement, Doc. No. HAU-D-056-005, Revision 1; Instructions for Continued Airworthiness, Doc. No. HAU-D-056-006, Revision 1; the paragraph(s) of a later amendment: the following paragraph(s) of a later...</td>
<td>Master Document List, Doc. No. HAU-D-056-001, Revision 1; Installation Manual, Doc. No. HAU-D-056-004, Revision 1; Flight Manual Supplement, Doc. No. HAU-D-056-005, Revision 1; Instructions for Continued Airworthiness, Doc. No. HAU-D-056-006, Revision 1; the paragraph(s) of a later amendment: the following paragraph(s) of a later...</td>
<td>Limited to serial number 2082, 2123 and 2468 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
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<td>Cabin Configuration and Refurbishment of the previously installed Private Lounge, Main Lounge and Upper Deck Cinema Area. Upgrade of the existingIFE System from eX2 to eX3 (Panasonic) and installation of a new Flight Information System.</td>
<td>The Type and OIS Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS.25.1567 for Amendment 12; CS25.603, CS25.788; CS25.851; CS25 Appendix 5; S25.1; S25.10; S25.11; S25.30.5.</td>
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<td>The Type and OIS Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS.29 Amendment 6 for the following requirement: 29.351; CS.29 Amendment 7 for the following requirement: 29.853. The requirement: 29.351 was deleted.</td>
<td>Certification Programme (CP) ref. AHSE-72244 Revision D dated02/02/2022 Flight Manual Supplement (FMS) for TrakkaBeam Searchlight installation: EASA-25.327 Revisiedated 2004/2022 Product Change Declaration (PCD) ref. AHSE-72238 Revision E dated01/2022 Instructio n for Continued Airworthiness (IAC).</td>
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<td>A300 B2-320</td>
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<td>Rev. level</td>
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<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10079252</td>
<td>SOCIETE AIR FRANCE S.A.</td>
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<td>A330-243</td>
<td>A330 Cabin RearrangementCabin Layout Change introducing a full Economy Seating Layout</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>The change is defined with revision 0 of Air France Change ApprovalSheet document EAS- A330-22-001. The installation shall be accomplished in accordance with revision 0 of the France Service Bulletin document AF5B - A330-22-001-35-001. The aircraft shall be maintained in accordance with revision 0 of Air...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/06/2022</td>
<td>Active</td>
<td>10079252</td>
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<td>10079262</td>
<td>NEPTUNE AVIATION SERVICES, INC.</td>
<td>FAA STC ST096966E</td>
<td>AVRO 146-RJ100</td>
<td>Installation of Honeywell Flight Data Acquisition Management System (FDAMS) with Flight Control Input Position Sensors</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document Lot, NER-MDL-P054701, Revision A, dated 27 June 2020; FAA approved revision instructions for Continued Airworthiness (ICA) for Installation of Honeywell Flight Data Acquisition Management System (FDAMS) with Flight Control Input Position Sensors, ICA-P254707, Revision NC, date...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/05/2022</td>
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<td>EASA STC no.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>10070265</td>
<td>AEROSPACE DESIGN &amp; COMPLIANCE LLC</td>
<td>FAA STD 8A4044NY</td>
<td>B300, B300C, B200CQ, B200O, B200CQ 200, 200C, A200</td>
<td>Installation of Upper SATCOM Radar. This modification 62 PAX LOPA</td>
<td>The Certification Basis for the original product remains applicable to the certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC 8A4044NY. The requirements for environmental protection and the associated requirements of noise and/or emissions levels of the aircraft are unchanged and remain applicable to this certificate/ approval...</td>
<td>1. Aerospace Design &amp; Compliance, LLC., Master Data List, B300-MLD-0029. Revision IR, dated 25 January 2021, or later FAA approved revisions. 2. Aerospace Design &amp; Compliance, LLC., Instructions for Environmental Protection (ICA), Document 264-80, Revision 0, dated 11 January 2021, or later FAA approved...</td>
<td>This STC is applicable to all serial numbers for the Make: Textron Aviation Inc. Models: 7300, 7300, 7300C &amp; 7300C aircraft with 15,000 lbs MTDW or PTA 60 Engines and Models: 200Q, 200Q, 200Q, 200Q, B200/B200C &amp; B200C/B200CST. This approval is limited to only the installation of the Radome forSatCom Antenna provisions...</td>
<td>17/05/2022</td>
<td>Active</td>
<td>10070265</td>
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<td>10070268</td>
<td>THEILE, THOMAS</td>
<td>PK-20 E</td>
<td>PK-20 - inst of winglets</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:The following paragraph(s) at a later amendment: CS-22 at Amdt. 2, 22.143 through 22.229, 22.375, 22.629, and 22.1581 through 22.1589.The requirements for environmental protection and the associated requirements of noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval...</td>
<td>Manufacturing drawings: TT-PK-20E WL-57-200, sheets 1-2 (installation instructions, TT-PK-20E WL-41, Rev. 2) dated May 2022. Supplement Flight Manual PK-20E Operation with winglets, Issue 1, Rev. 0, dated March 2022. Two later revisions of the above listed documents approved/ accepted/under the EASA...</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point A.109.</td>
<td>17/05/2022</td>
<td>Active</td>
<td>10070268</td>
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<tr>
<td>10070302</td>
<td>AERODATA AG</td>
<td>B300</td>
<td>Activation of the FLARM system</td>
<td>The Type and OSD Certification Bases (CB) for the original product are unchanged and remain applicable to this certificate/ approval. The requirements for environmental protection and the associated requirements of noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval...</td>
<td>Design Definition Document MLD-AIC-110, rev. 5. Airplane Flight Manual Supplement AIC-182, Rev. 2. Installation Instructions for Continuous Airworthiness (ICA), Document 26A-80, Revision 1, dated 18 May 2022. Supplement Flight Manual PK-20E Operation with winglets, Issue 1, Rev. 0, dated March 2022. Two later revisions of the above listed documents approved/ accepted/under the EASA...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair and any emissions levels of the product.</td>
<td>19/05/2022</td>
<td>Active</td>
<td>10070302</td>
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<td>10070311</td>
<td>RECAERO</td>
<td>DHC-6 SERIES 300</td>
<td>Ladder relocation from aft door LH to aft door RH</td>
<td>The Type and OSD Certification Bases (CB) for the original product are unchanged and remain applicable to this certificate/ approval. The requirements for environmental protection and the associated requirements of noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval...</td>
<td>Design Service Bulletin document RCP026DSB001 Issue 3. Flight Manual Supplement document RCP026FMS001 Issue 2/3. Aircraft Manual Supplement document RCP026AMS001 Issue 2/v. Two later revisions of the above listed document(s) approved/accepted/under the EASA...</td>
<td>Applicable to MSN 292, 298, 300, 730, 745. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair and any emissions levels of the product.</td>
<td>19/05/2022</td>
<td>Active</td>
<td>10070311</td>
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<td>10070328</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>737-700</td>
<td>62 PAX LOPAThis modification introduces a cabin layout using 31 business class/doubtly seats generating a 62 passenger seating configuration</td>
<td>The Type and OSD Certification Bases (CB) for the original product are unchanged and remain applicable to this certificate/ approval. The requirements for environmental protection and the associated requirements of noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval...</td>
<td>The change is defined with revision 0 of GVH Aerospace Master Data List document GVH-13103-01-MLD-0016a 1The aircraft shall be maintained in accordance with Issue 1 of GVH/Aerospace Instructions for Continuous Airworthiness documentGVH-13103-01-ICA-01. Two later revisions of the above listed document(s) approved/accepted/under the EASA...</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair and any emissions levels of the product.</td>
<td>20/05/2022</td>
<td>Active</td>
<td>10070328</td>
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<td>STC Holder</td>
<td>Description</td>
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<td>J &amp; C AEROSPACE DESIGN AND CERTIFICATI ON SERVICES EEU Ltd.</td>
<td>Cabin with 189 PAX seating arrangement and adaptation of the passenger cabin to 189 PAX to fulfill operational requirements, by replacement of existing ETSO-C127a seats with new Geven Exteriors.</td>
<td>Active</td>
<td>JET AVIATION AG, AEROLITE AG</td>
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<td>EASA STC no.</td>
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<td>Revision</td>
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<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10079381</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>737-800</td>
<td>FAA SAA1871CH</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>GVH-13130-01-MOD-01 - 186Y PAX LOPA/This modification introduces a cabin layout using 62 economy class/seat/ seats generating a 186 passengers seating configuration.</td>
<td>The Type and OSD Certification Bases (CB) for the original product/production are applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>The change is defined with Issue 1 of GVH Aerospace Master Data/Document GVH-13130-01-MDL-01, dated 25/MAY/2022/ The aircraft shall be maintained in accordance with Issue 1 of GVH Master Instructions for Continued Airworthiness document GVH-13130-01-ICA-01, dated 25/MAY/2022/ for later revisions of the above listed document(s).</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change/repair and or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>31/05/2022</td>
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<td>10079382</td>
<td>SMR ACQUISITION LLC dba SMR TECHNOLOGIES, INC</td>
<td>1900, 1900C, 1900D</td>
<td>FAA SAA1871CH</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>Replacement of certain Goodrich pneumatic de-ice boots</td>
<td>The Type and OSD Certification Bases (CB) for the original product/production are applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Pneumatic De-Icers Beech 1900 Series, SMR517, Rev. 6, dated 28/Sep/2022/ Master Drawing List for Beech 1900, 1900C, 1900D, MDL-004, Rev. G dated 13/Sep/2022/ Master Drawing List for Beech 1900D, MDL-001 Rev. J dated 17/Sep/2022/ for later revisions of the above listed document(s).</td>
<td>Approval of this change in type design extends to the applicable model(s) with previously approved de-ice boots/intermix of these de-ice boots with previously approved de-ice boots permitted/ Prior installation of this change/repair is mandatory/ between this change/repair and any other previously installed changes/repair.</td>
<td>31/05/2022</td>
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<td>10079395</td>
<td>UAVIONIX CORPORATION</td>
<td>FAA STC SAA4568CH</td>
<td>FAA STC SAA4568CH</td>
<td>UAVIONIX CORPORATION</td>
<td>Installation of uAvionix tailBeaconX, a tail mounted position light with integrated Automatic Dependent Surveillance-Broadcast (ADS-B) Out. 1090MHz Mode-S transponder functionality</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation instructions as listed on AML No. SA04568CH.</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change/repair and or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>02/06/2022</td>
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<td>10079395</td>
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<td>10079396</td>
<td>CAVOK</td>
<td>208B</td>
<td>FAA STC SAA1871CH</td>
<td>UAVIONIX CORPORATION</td>
<td>Cessna 208B Cabin layout 4 Passengers</td>
<td>The Type and OSD Certification Bases (CB) for the original product/production are applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Dossier de Certification, document 208-089-CTF-003 Issue 01/June/2022/Supplement to Pilot Operations Handbook, document 208-089-HBR-002 Issue 01/June/2022/ for later revisions of the above listed document(s).</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change/repair and or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>02/06/2022</td>
<td>Active</td>
<td>10079396</td>
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<td>10079401</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>A320-253N-271N-272N-272N-274N-275N</td>
<td>FAA STC SAA1871CH</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>Airbus A320 - 180 PAX LOPA</td>
<td>Installation: This design change installs new economy class seats to accommodate a 19 passengers’ layout on Airbus A320. The seats to be installed are economy class seats and do not contain any electrical equipment. Therefore no seat, seat-to-seat or cabin wiring is installed.</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: (JAR 25.1321(a)) Delete Change 13. The requirements for environmental protection and the associated certification noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List ref: GVH-11907-01-MDL-01, Issue 1 dated 26/SEP/2022/ Certification Programme: GVH-11907-01-CP-01 Issue 0 dated 03/DEC/2020: Instructions for Continued Airworthiness: GVH-11907-01-ICA-01 Issue 0 dated 26/Apr/2022/ for later revisions of the above listed document(s).</td>
<td>Following changes must be installed concurrently:- GVH-69999-17-MOD-01/ Installation of Interior Placards- GVH-10802-02-MOD-01/ Installation of Emergency &amp; Safety Equipment- EASA STC 100/4013/ Luftansa Technik AG - Installation ohm-electrical Floor Path marking system/ Prior to installation of this change/repair and any other previously installed change/repair and or repair will not introduce any adverse effect upon the airworthiness of the product.</td>
<td>03/06/2022</td>
<td>Active</td>
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<td>10079404</td>
<td>TRANSPORTES AEREOS PORTUGUESES, S.A.</td>
<td>A330-041 A330-041</td>
<td>FAA STC SAA1871CH</td>
<td>TRANSPORTES AEREOS PORTUGUESES, S.A.</td>
<td>TAP-ACO300: ATA 25 - Bucher stretcher Installation on TAP fleet</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: (JAR 25.561(d), 25.785(a), 28.182(a), 25.807, 25.812(d), 28.820, 25.812(e), 25.815, 25.1383(x)(a), 25.1380(a), 28.141) Certification Report TAP-AC0200-01R1 dated 01/Apr/2022/ for later revisions of the above listed document(s).</td>
<td>EASA STC no.</td>
<td>03/06/2022</td>
<td>Active</td>
<td>10079404</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>AKADEMISCHE FLEGERGRUPPE BERLIN</td>
<td>GROSS or 103 &quot;TWIN II&quot;</td>
<td>Installation of a wing-mounted sleeve with closed cross-section to carry measurement equipment for inflight measurements.</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval without any impact.</td>
<td>None</td>
<td>09/06/2022</td>
<td>Active</td>
<td>10079432</td>
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<td>FOKKER SERVICES B.V.</td>
<td>B737-700, 737-800ER</td>
<td>CRR-0435 2</td>
<td>Certification Program ref. CRR-0435 issue 7 - Compliance Record Report ref. CRR-0435 issue 2 - AFM Supplement ref. B737-SLS-ASM-S-001 issue 3 - MMEF Supplement ref. B737-SLS-AMEL-S-001 revision 4 - FCD Supplement ref. B737-SLS-FCD-S-001 issue 40 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This STC is to be applied on aircraft in conjunction with EASA STC 10074404 “Activation of standalone GNSS sensor system on Boeing 737-NGAircraft”, ref CRR/ CRR-0425, installing a CMA-S04 GPS Landing System Sensor Unit (GLSSU).Prior to installation of this change/repair it must be determined that the aircraft is equipped with latest revision of the above listed document(s) approved/accepted under the EASA system.</td>
<td>18/07/2022</td>
<td>Active</td>
<td>10079451</td>
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<td>FOKKER SERVICES B.V.</td>
<td>B737-700, 737-800ER</td>
<td>CRR-0425 1</td>
<td>Certification Program ref. CRR-0435 issue 7 - Compliance Record Report ref. CRR-0435 issue 2 - AFM Supplement ref. B737-SLS-ASM-S-001 issue 3 - MMEF Supplement ref. B737-SLS-AMEL-S-001 revision 4 - FCD Supplement ref. B737-SLS-FCD-S-001 issue 40 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This STC is to be applied on aircraft in conjunction with EASA STC 10074404 “Activation of standalone GNSS sensor system on Boeing 737-NGAircraft”, ref CRR/ CRR-0425, installing a CMA-S04 GPS Landing System Sensor Unit (GLSSU).Prior to installation of this change/repair it must be determined that the aircraft is equipped with latest revision of the above listed document(s) approved/accepted under the EASA system.</td>
<td>18/07/2022</td>
<td>Active</td>
<td>10079451</td>
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<td>FIELD AVIATION Company Inc.</td>
<td>DHC-8-102</td>
<td>The change describes an aircraft reconfiguration to restore the certification configuration to the DHC-8-102, S/N 038 aircraft after operation by Department of National Defense (DND).</td>
<td>The Type and OSD Certification Bases (CS) for the original product remain applicable to this certificate approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval.</td>
<td>Master Document list MLD- F260-21-01F edition 5 dated 10.06.2022 aircraft flight manual supplement SUPMF-260-21-01F dated 10.06.2022 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>14/06/2022</td>
<td>Active</td>
<td>10079471</td>
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<td>ATELIER CHABORD</td>
<td>F260E, F260F</td>
<td>The Type and OSD Certification Bases (CS) for the original product remain applicable to this certificate approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval.</td>
<td>The change is defined with FAA DDL-1994 Rev B(&gt;)+&lt;n&gt;7&lt; rev revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation/Procedures of EU Canada Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined if the aircraft is equipped with latest revision of the above listed document(s) approved/accepted under the EASA system.</td>
<td>14/06/2022</td>
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<td>10079474</td>
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<td>PANKL AEROSPACE SYSTEMS EUROPE GMBH</td>
<td>A320-214</td>
<td>The change is defined with revision 1 of Pankl Certification Program Document CER_22004_04302022Th e installation shall be accomplished in accordance with revision 0 of Pankl Instructions for Continuous Airworthiness Document/CA 20004 070 42022 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The modification is only applicable to the basic aircraft model without MOD defined by EASA TCDS. EASA is to be determined. Prior to installation of this change/repair it must be determined that the aircraft is equipped with latest revision of the above listed document(s) approved/accepted under the EASA system.</td>
<td>15/06/2022</td>
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<td>10079478</td>
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<td>JET AVIATION AG</td>
<td>B777 KA Band Antenna STC Installation of KA Band Satcom System</td>
<td>B777 KA Band Antenna STC Installation of KA Band Satcom System</td>
<td>The change is defined with revision 1 of Pankl Certification Program Document CER_22004_04302022Th e installation shall be accomplished in accordance with revision 0 of Pankl Instructions for Continuous Airworthiness Document/CA 20004 070 42022 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the aircraft is equipped with latest revision of the above listed document(s) approved/accepted under the EASA system.</td>
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<td>JET AVIATION AG</td>
<td>10079510</td>
<td>REV. 2</td>
<td>777-200UR</td>
<td>UBLA MA137600-001 Issue A - Declaration of Compliance/BIL L-01276CCS-01 Issue 3 - Certification Compliance Sheet/BIL</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/06/2022</td>
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<td>ARCHEON HOLDINGS, L.L.C.</td>
<td>10079514</td>
<td>REV. 1</td>
<td>FAA ST04518AT</td>
<td>737-9 737-800 737-8 737-700 737-400 737-300</td>
<td>Installation of an Avionica, avWiFi wireless system with Cyber Security/Compliance and Hosted Application Functionality on B737-300/400/700/800/8/9.</td>
<td>FAA approved Airplane Flight Manual Supplement for ARCHEON Holdings LLC, 20450-AFMS, Revision IR. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>21/06/2022</td>
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<td>FOKKER TECHNIEK B.V.</td>
<td>10079530</td>
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<td>VVIP Cabin Refurbishment</td>
<td>737-800</td>
<td>VVIP Cabin Refurbishment</td>
<td>Master Data List B737-737- HSMVS-AFMS-9-001 Issue 2AFM Supplement B737-737- HSMVS-AFMS-9-001 Issues 3Instructions for continued Airworthiness B737-737- HSMVS-ICA-S-002 Issue 2AMIS Supplement B737-737- HSMVS-MMEL-S-001 Issue 2for later revisions of the above listed document(s) approved/accepted on behalf of EASA in...</td>
<td>21/06/2022</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>10079537</td>
<td>FAA STC ST04330AT-D</td>
<td>GV-SP G-GV GIX GV</td>
<td>Installation of Personal Electronic Device (PED) Interface on all Gulfstream G-IV, GV, GIV-X and GV-SP aircraft.</td>
<td>The instructions contained in the Installation Interface and Limitation for PEDS STC Compatibility with Follow-On Aircrafts CE515718012, Revision (-) dated February 25, 2020 or later approved revision must be followed for subsequent installations of this STC. Prior to installation of this change/repair...</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>10079539</td>
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<td>A321-251NX</td>
<td>Supplemental Type Design Document: STDD00-20-1YM Issue 1400/Airline Flight Manual: A-00-2045AFM Issue 03/05/Cabin CrewData: A-00-2045-CCD Issue: 0200/Flight Crew Data: A-00-2045-FCD Issue: 0200/Flight Crew Information: A-00-2045- FCI Issue: 04Instructions for Continued Airworthiness A-00-2045 Issue: 0.</td>
<td>Aerospace modified per this STC are considered to be a &quot;new type&quot; subject to A321-200 commercial configurations. Cabin crew operating Aerospace modified per this STC must complete aircraft type specific training. In accordance with EC-125, which must include the approved OSD/Cabin Crew Data (CCD) Prior...</td>
<td>21/06/2022</td>
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<td>JET AVIATION AG</td>
<td>STC Holder</td>
<td>777-200LR</td>
<td>Head of State Cabin CompletionInstallation of a Head of State cabin interior into a green BOEING777-200LR</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:: - Special Condition(s): OR F-6 Security Protection of Aircraft Systems and Networks: The following paragraph(s) at a later amendment CS25:24.25, 25.27...</td>
<td>STC: M-0305GD-IN-001 ISSUE G MASTER DATA LIST.JSBSC: M-0305FAJ-001 ISSUE C AIRPLANE FLIGHT MANUAL SUPPLEMENT:JSBSC: M-0305GCA-001 ISSUE A INSTRUCTIONS FOR CONTINUED AIRWORTHINESS:JSBSC: M-0305GCS-001 ISSUE G CERTIFICATION COMPLIANCE SHEET:JSBSC: M-0305GML-001 ISSUE B MASTER MINIMUM EQUIPMENT LIST SUPPLEMENT:</td>
<td>- Private Operation (Not-for-Hire, Non-Commercial): - Non-Smoking Aircraft. Max occupancy: 98 Pax, 4 Flight crew, 11 Cabin Crew: The aircraft must be fitted with Boeing Delivery Interior ArrangementDiagram - Zero Occupancy Certification, drawing reference LOFA 772-226Rev D - Exterior Exit Marking - air...</td>
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<tr>
<td>JET AVIATION AG</td>
<td>REV. 2</td>
<td>A319-153N</td>
<td>VIP Cabin Interior CompletionThe modification installs a full VIP interior to a green ACJ A319-153N aircraft. The aircraft must be fitted with Boeing Delivery Interior ArrangementDiagram - Zero Occupancy Certification, drawing reference LOFA 772-226Rev D - Exterior Exit Marking - air...</td>
<td>The Certification Basis for the original product/programme applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval w...</td>
<td>JBSC-M-03049 MDL-001 issue G dated 27 Oct 2022 JBSC-M-03049 MAF-001 issue A dated 27 Oct 2020JBSC-M-03049 AFM-001 issue B - Aircraft Flight Manual Supplement dated 21 June 2022JBSC-M-03049 CCD-001 issue A - Cabin Crew Data dated 31 May 2022JBSC-M-03049 ICA-001 issue B - Instructions for Continu...</td>
<td>- Private Operation (Not-for-Hire, Non-Commercial): - Non-Smoking Aircraft. Max occupancy: 98 Pax, 4 Flight crew, 11 Cabin Crew: The aircraft must be fitted with Boeing Delivery Interior ArrangementDiagram - Zero Occupancy Certification, drawing reference LOFA 772-226Rev D - Exterior Exit Marking - air...</td>
<td>31/10/2022</td>
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<td>S-TEC CORPORATION</td>
<td>FAA STC</td>
<td>505</td>
<td>S-TEC HeloSAS Stability Augmentation System installation.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:: - Special Condition(s): OR F-6 Security Protection of Aircraft Systems and Networks: The following paragraph(s) at a later amendment FAR 27.1(a)(1)(v)...</td>
<td>Master Data List No. ST-974-MDL-001, Rev. A dated 30 January 2020Rotorcraft Flight Manual Supplement ST-974-RFMS-001 Rev. P dated 23 June 2021for Continuance of airworthiness ST-974-ICA-001 Rev. C dated 30 October 2021Flight Crew Data ST-974-FCDS-001 Rev C dated 13 June 2022for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/06/2022</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>A321-271NX-272NX A321-271N-272N A321-251NX-252NX-253NX A321-211-212-213 A321-211-212 A321-211-212-213-214-215-216</td>
<td>IRIDIUM ANTENNA INSTALLATION - MECHANICAL PROVISIONS</td>
<td>The Type and OSD Certification Bases (CB) for the original product/programme applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval w...</td>
<td>MDL Doc AE200100MDL Rev. : (2006/02/22) ADAI-200100MDL Rev.: (04/01/2022)for later revision of the above listed document(s) approved/accepted under the EASA system.</td>
<td>ICA Doc AE200101ICA Rev. 0 (04/01/2022)Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/06/2022</td>
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<td>Date of Issue</td>
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<tr>
<td>ISRAEL AEROSPACE INDUSTRIES</td>
<td>Rev. 4</td>
<td>SA300</td>
<td>737-800</td>
<td>Conversion of a Boeing 737-800 PAX airplane configuration into a SpecialFreighter (SF). This conversion to a special freighter configuration among other features includes a main deck Class E cargo compartment and associated smoke detection system, installation of a main deck cargo door withoutstanding...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): D-02 “Cargo compartment fire protection of essential systems” issue 2; D-03 “Access to class...</td>
<td>The following IARs documents - 365-00-00- M4560 Master Document List MD, revision AA, and 365-00-00- M4565 Airplane Flight Manual Supplement (EASA) revision 1 and 365-00-00- M4568 Structural Repair Manual Supplement revision 8, and 365-00-00- M4544 Maintenance Planning Document (MPD) Supplement to...</td>
<td>29/01/2024</td>
<td>Active</td>
<td>10077517</td>
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<td>NSE</td>
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<td>EC 130 T2</td>
<td>AS 350 D</td>
<td>Installation of BCT (Communication and Tracking Box)</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the...</td>
<td>- Mounting Instruction MD-004002CAA A2: Rotorcraft Flight Manual Supplement RMMS-BCT- Excuse EN (11) for Continued Airworthiness ICA-000402CAA A2 for later revisions of the above listed document(s) approved/acceptedunder the EASA system.</td>
<td>28/06/2022</td>
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<td>GULFSTREAM AEROSPACE CORP</td>
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<td>FAA STC</td>
<td>GVI-G600</td>
<td>Installation of an Air Ionization System</td>
<td>The Certification Basis for the original product remains applicable unless certified approval, except where amended by additional or later amendments if indicated on FAA STC ST0434AT-D. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the...</td>
<td>1. Gulfstream Aerospace Corporation Index List GDL 154000059. Rev (1), dated 8 March 2022; 2. FAA approved Instructions for Continued Airworthiness AERONAVIG Amdt. 01 of April 2022, approved/accepted under the EASA system</td>
<td>28/06/2022</td>
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<td>J &amp; C AEROSPACE DESIGN</td>
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<td>A330-341/-342/-343</td>
<td>A330-321/-320/-323</td>
<td>Transportation of Cargo in the Cabin/This change covers passenger cabin conversion for transportation of cargo to comply with the operator's requirement to perform cargo flight on cabin floor without cargo pallets using cargo nets and restraint systems.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Requirement(s): CRI D-01 (Transportation of cargo in passenger compartment)The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>AFM Supplement: JCAERO/AFM/0714-1 Issue 1; Cabin and Cargo Handling Limitations manual: JCAERO/AFM/0714-1; JCAERO/AMM/0714-1 Issue 1 and JCAERO/AFM/0714-1 Issue 2; FAA approved Instructions for Continued Airworthiness - Aircraft Systems List GVI-G500/GVI-G500, GCT73782/001 1.00000059. Rev (1), dated 28 February 2022; 2. FAA approved Air Ionization System - EWIS - Instructions to...</td>
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<td>SIRIUM AEROTECH S.L.</td>
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<td>A340-311/-312/-313</td>
<td>A330-223/-243/-228F/-248F</td>
<td>ADS-B Upgrade for A330 and A340 Airplanes</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>SA20-235-MLD Rev. 02 SA20-235-AMS Rev. 02 SA20-235-AMS Rev. 02</td>
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<td>NORTHWEST AEROSPACE TECHNOLOGIES INC.</td>
<td>REV. 1</td>
<td>FAA ST07308SE</td>
<td>767-9</td>
<td>Installation of a VIP interior in accordance with Greenpoint Technologies, Inc., Drawing List G14561003-DL-01 Revision 1: Transfer of the STC 10079604 from Greenpoint Technologies, Inc. to Northwest AEROSPACE TECHNOLOGIES, Inc - Editorial correction of EASA Certification Basis.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - Special Condition(s): (J) CRF D-02 Size facing seats, issue 2 dated 28 June 2021: CRF F-01 Information security protection of aircraft systems and networks.</td>
<td>Greenpoint Technologies, Inc. Master Drawing List G14561003-DL-01 Revision FGreenpoint Technologies, Inc. Airplane Flight Manual Supplement G14561003- FM-02 Revision FGreenpoint Technologies, Inc. Minimum Equipment List Supplement G14561003- MEL-01 RevGreenpoint Technologies, Inc. Supplement...</td>
<td>Aeroplanes modified in accordance with this STC are a variation of a currently approved STC(s) for other commercial configurations. Cabin crew operating the aeronautical equipment per this STC must complete an aircraft difference training in accordance with EU OPS ORR CC.130 which must be approved EASA/Cabin Crew Data – Supp...</td>
<td>08/10/2022</td>
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<td>SIERRA INDUSTRIES Ltd.</td>
<td>10079609</td>
<td>FAA ST 53735W</td>
<td>T210M, P210N, 210K, 210L, 210M, TM210K, TM210L</td>
<td>Remove main landing gear doors and associated hydraulic system, and cover the wheel wells with a fairing.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Installation Instructions dated September 12, 1984. The above listed document(s) approved/accepted under EASA in accordance with the Technical Implementation Procedures of E.U. USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>SAA SOLUTIONS FOR AVIATION, S.L.</td>
<td>330-243</td>
<td>B300</td>
<td>Skytrack Antenna installation/ removal</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/ remain applicable to this certificate/ approval.</td>
<td>Master Document List P2F-00040-MDL-000 Rev 01/Instructions for continued airworthiness P2F-00040- ICA Rev 01/Revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/07/2022</td>
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### List of EASA Supplemental Type Certificates

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<th>Associated Technical Document</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
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<td>PMV ENGINEERING</td>
<td>REV. 1</td>
<td>ATR 72-212 A</td>
<td>ATR 72-211, ATR 72-212, ATR 72-201, ATR 72-202</td>
<td>P0F-0040 Window plugs ATR72REV. 1: removing the windows inspection ICA task</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/ remain applicable to this certificate/ approval.</td>
<td>Master Document List P2F-00040-MDL-000 Rev 01/Instructions for continued airworthiness P2F-00040- ICA Rev 01/Revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/01/2023</td>
<td>Active</td>
<td>10079622</td>
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<tr>
<td>CAVOK</td>
<td>B300</td>
<td></td>
<td></td>
<td>Skytrack Antenna installation/ removal</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/ remain applicable to this certificate/ approval.</td>
<td>Service Bulletin, document 350-116-BB-007, Issue 0 dated 29 November 2022 – Revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship with this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/07/2022</td>
<td>Active</td>
<td>10079645</td>
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<tr>
<td>SABENA TECHNICS BSC</td>
<td>A321-271N-372N</td>
<td>A321-271N-25SN-25IN</td>
<td>A321-2511-222</td>
<td>A321-2511-222</td>
<td>A321-2712-212-213</td>
<td>A321-1111-1121-131</td>
<td>Cabin Reconfiguration from 195PAX (128C + 183 Y/C) to 220PAX (220 Y/C)</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/ remain applicable to this certificate/ approval.</td>
<td>Master Data List. Ref: 41006-01-A-MDL-R00 dated July 14th 2022/Instructions for continued airworthiness, Ref: 41006-01-A-MDL-R00 dated July 14th 2022/Revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>As a concurrent requirement of this change, accomplishment of this STC requires the simultaneous accomplishment of the following: Modification Installation of a photo-luminescent EPPM as per Luftansa Technik/Engeneering Bulletin A320-EB33-0384, approved by EASA STC No. 10014013/Prior to installation...</td>
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<tr>
<td>STC Holder</td>
<td>STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10079656</td>
<td>STC Holder</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>A320-230, A320-243, A330-201, A330-202, A330-203</td>
<td>Passenger Service System &amp; In-seat USB Power Installation.Installation of Inflight Canada(FC) passenger service system with In-seat USB power system (PSS+HP); This change includes the installation of the PSS+HP equipment, electrical harnesses and connection to an in-cabin installation of the seat moun...</td>
<td>The Certification Basis for the original product as amended (modify/addition or alternative airworthiness requirements).The following paragraph(s) at a later amendment CS-25 Initial Issues: 20.1.23; 25.27; 29.381; 303.307; /337.341; 365/; 651.652/; 671/; 693/; 695/; 607/; 613/; 625/; 641/</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/07/2022</td>
<td>Active</td>
<td>10079656</td>
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<tr>
<td>10079658</td>
<td>STC ref.</td>
<td>LEE AEROSPACE, INC.</td>
<td>FAA ST04208NY</td>
<td>Installation of LAI Windshields</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval, except where amended by additional or later amendments as indicated on FAA STC ST04208NY.The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/07/2022</td>
<td>Active</td>
<td>10079658</td>
<td></td>
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<tr>
<td>10079659</td>
<td>Advanced Aerospace Developments OMB</td>
<td>MBB-BK117 C-2</td>
<td>Installation of &quot;7EOS&quot; Hanseß ARGOS-II HD Electro-Optical System(EOS)</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval/w...</td>
<td>Certification Program, Document No. AAD-CP-07EOS-001 Issue 03 dated 24.06.2022 Master Drawing List, Document No. AAD-DM-07EOS-001 Issue 07 dated 03.06.2022 Master Documents List, Document No. AAD-ML-07EOS-001 Issue 01 dated 03.06.2022 Flight Manual Supplement, Document No. AAD-FMS-07EOS-001 Issue 022.</td>
<td>This STC is not approved for models BMB-BK117 C-2e. The installation of the &quot;7EOS&quot; Harssh ARGOS-II HD Electro-Optical System is not equipped with a video device and requires prior installation of the utility step mount from AIRFILM CAMERA SYSTEMS, INC.(STC No. SR02137LA). It is the operator's responsibility...</td>
<td>07/07/2022</td>
<td>Active</td>
<td>10079659</td>
<td></td>
</tr>
<tr>
<td>10079660</td>
<td>FAA ST03358NY</td>
<td>BELL TEXTRON, Inc.</td>
<td>Installation of interior blankets in passenger compartment</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval/w...</td>
<td>Installation Drawing List Report No. AA-12055 Revision D, dated 19 September 2013,awson Aerospace LLC Instructions for Continued Airworthiness Report No.AA-12055 Revision C,dated 30 May 2013. Prior to installation of the product.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/07/2022</td>
<td>Active</td>
<td>10079660</td>
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<tr>
<td>10079682</td>
<td>J &amp; C Aerospace Design And Certification Services EUU Ltd.</td>
<td>A340-311-312-313, A340-211-212-213</td>
<td>Transportation of Cargo in the Cabin/This change covers passenger cabin conversion for transportation of cargo to comply with the operator's requirement to perform cargo flightplan cabin floor without cargo pallets using cargo nets and restraintstrips</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise a...</td>
<td>AFM Supplement: JCAERO/AFM/0773-1 Issue 1.0 Cabin and Cargo Handling limitations manual: JCAERO/Supp/0773-2 Issue 1.0AFM Supplement: JCAERO/AIM/0773-1 Issue 1.0Master Data List JCAERO_MDL_0773, ISS 1.0MPD supplement: Page 7 and 8 JCAERO_MDCD_0773 Issue 1.0 prior revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures o...</td>
<td>Aeroplanes modified by this STC must be maintained in accordance with - JCAERO/AIM/0773-1 and Page 7 and 8 JCAERO_MDCD_0773 Issue 1.0 and JCAERO/Supp/0773-2 Issue 1.0. The allowance of transport cargo in passenger compartment is limi...</td>
<td>08/07/2022</td>
<td>Active</td>
<td>10079682</td>
<td></td>
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<tr>
<td>STC Holder</td>
<td>Foreign/NAACertificate/Registration</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>THALES AVIONICS Inc.</td>
<td>FAA STC, ST0446A</td>
<td>A321-272N, A321-271N, A321-253N, A321-252N, A321-251N, A321-232, A321-211, A321-212, A321-272N, A321-273N, A320-251N, A320-252N, A320-253N, A320-212, A320-213, A320-214, A320-215</td>
<td>Modification of existing Thales Avionics TopFares In-flight Entertainment (IFE) System</td>
<td>The Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS.25.Amdt.25,305 (a),(b),(c) 25.30725,56153,61253,62583,853 (a),25.859(a),(25.13511(a)1) The requirements for environmental p...</td>
<td>- FAA Approved Thales Avionics IFE System, Drawing No. 255274</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/07/2022</td>
<td>Active</td>
<td>10079593</td>
<td></td>
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<tr>
<td>LUTHANSA TECHNIK AG</td>
<td></td>
<td>A340-213</td>
<td>Antenna Sys. Installation &amp; Activation of JU/LINK</td>
<td>The Type and OSD Certification Bases (CB) for the original product produces/aircraft applicable to this certificate/approval. The requirements for environmental protection and the associated/controlled noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval w...</td>
<td>Supplemental Type Design Definition STDD03-006M Issue 02 Rev 00/Flight Manual Supplement A-00-2126/AFM issue01 Rev 00/Instructions for Continued Airworthiness A-00-2126/ICA issue 01 Rev 00/ later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>11/07/2022</td>
<td>Active</td>
<td>10079651</td>
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<tr>
<td>ATLANTIC AVIATION GROUP LIMITED</td>
<td></td>
<td>737-800</td>
<td>Avolon B737-800 (MSN 35646) LOPA Recomfigurion</td>
<td>The proposed change is a cabin reconfiguration from 138 to 169 passenger layout. The following modifications will be performed: Removal of existing (183 Pax) shiftpack of seats, Installation of new (199 Pax) shiftpack of seats R5030</td>
<td>Master Data List AAG-22- C004-MDL, Revision 3, dated 10/2022</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/07/2022</td>
<td>Active</td>
<td>10079650</td>
<td></td>
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<tr>
<td>MT-PROPELLER ENTWINING GmbH</td>
<td></td>
<td>E33, F33, HS, K35, K36, M35, 35-830, 35-353, 35-33, 35-33, 35-33, 35-33</td>
<td>Installation of MTV-9-D200-52</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: 25.1707(0)(x)(a), 25.1707(0)(b), 25.1709, 25.1711, 25.1712(a)(c), 25.1715, 25.1717, 25.1719, and 25.1720 at CS.2</td>
<td>According to MT-Propeller Master Data List No. E-3776 Rev. 0/0 To be operated in accordance with MT-Propeller AFM Document No. E-3779Rev. 0/0 To be installed i.a.w. MT-Propeller Installation Instruction Doc. No.E-3779 Rev. 0/0 To be maintained i.a.w. MT-Propeller ICA Doc. No. E-3780 Rev. 0/0 later...</td>
<td>12/07/2022</td>
<td>Active</td>
<td>10079703</td>
<td></td>
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<tr>
<td>MT-PROPELLER ENTWINING GmbH</td>
<td></td>
<td>414</td>
<td>Extension of the existing LBA STC SA 1318 Installation of MTV-14-D8-C-F/CF188-30</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS.36 Amdt. 6 FAR 33AMdt 36-31This certificate/approval involves a change with an impact on the noise database...</td>
<td>According to MT-Propeller Master Data List No. E-3964 Rev. 0/0 To be operated in accordance with MT-Propeller AFM Document No. E-3965Rev. 0/0 To be installed i.a.w. MT-Propeller Installation Instruction Doc. No.E-3964 Rev. 0/0 To be maintained i.a.w. MT-Propeller ICA Doc. No. E-3965 Rev. 0/0 later...</td>
<td>12/07/2022</td>
<td>Active</td>
<td>10079705</td>
<td></td>
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<tr>
<td>SAFRAN AEROSYSTEMS</td>
<td></td>
<td>EC 130 T2 EC 130 B4</td>
<td>Quick Removal System for Airbus Helicopters (AH) Emergency Floatation/Syste</td>
<td>The Type and OSD Certification Bases (CB) for the original product produces/aircraft applicable to this certificate/approval. The requirements for environmental protection and the associated/controlled noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval w...</td>
<td>Master Drawing List GT-0017426 issue 0/0 Aircraft Maintenance Manual Supplement ILS-000235 issue 03 Certification Program GT-00017181 Rev. 0/0 later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>This STC can be applied to AH EFS configuration (P/N 704A2660004 for LH5 and P/N 704A2650003 for RH5) Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/07/2022</td>
<td>Active</td>
<td>10079715</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>10079719</td>
<td>GLENERT + KOHLMETZ GMBH</td>
<td></td>
<td></td>
<td>DO-800, S DO-600, 18 DO-600, DO-300, ELAN, ACRO DO-300, ELAN DO-300, CLUB ELAN DO-300</td>
<td>Installation of the turbine PSR T-01 into DG-300.</td>
<td>The type certification bases for the original product remain applicable. Due to the installation of the turbine, the following paragraphs are additionally affected: JAR 22.901 - JAR 22.903 - JAR 22.905 - JAR 22.909 - JAR 22.965 - JAR 22.997 - JAR 22.917 - JAR 22 9732...</td>
<td>- Englanders Flughandbuch, Ausgabe Dezember 2021.</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/07/2022</td>
<td>Active</td>
</tr>
<tr>
<td>10079731</td>
<td>ROEDER PRAEZISION GMBH</td>
<td></td>
<td>B300, B300C</td>
<td></td>
<td>Infrared Suppressing Systems (IRSS)</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Installation Instructions, document A-1915-001, Rev 10A rev. List, document 21-06276L, Rev BAircraft Flight Manual Supplement, document 21-06276FM Rev Chermation for Continued Airworthiness, document 21-06276CA Rev B or later revisions offe...</td>
<td>Applicable to King Air B300/B300C aircraft with PT6A-67A engines. MT-275-bladed propeller and up to 8000kg (17,637 lb.) MTOW. Prior to installation of this design change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/07/2022</td>
<td>Active</td>
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<tr>
<td>10079734</td>
<td>HELLIONE (NORWAY) AS</td>
<td></td>
<td>AS 332 L1</td>
<td></td>
<td>Night Vision Imaging System (NVIS) re- evaluation (ref. modification/AHSE-72950).</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 29 Amend 2 for the following requirements: 29.1, 29.141, 29.1383, 29.1501, 29.1529, 29.1561, 29.1563, 29.15...</td>
<td>Certification Programme (CP) ref. AHSE-72952 Revision B, dated 26/01/2022. Flight Manual Supplement (FMS) for NVIS re-evaluation ref. HTNo. 3.361Revision dated 14/07/2022 Product Change Declaration (PCD) ref. AHSE-72950 Revision A, dated 01/02/2022. Instructio...</td>
<td>This design change is limited to the AS 332 L1 Serial Numbers (S/N) 3042, 3043, 3045 and 3047 with configuration described in the Project Description document ref. AHSE-72956 Revision A dated 26/01/2022 and ICAdocument ref. AHSE-72958 Revision C dated 13/06/2022. Rolonet modified by this design cha...</td>
<td>18/07/2022</td>
<td>Active</td>
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<tr>
<td>10079735</td>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td>REV. 2</td>
<td>ERJ 190-300, ERJ 190-400 ERJ 190-500 ERJ 190-200 ST ERJ 190-200 LR ERJ 190-100 NWG ERJ 190-100 STD ERJ 190-100 SR ERJ 190-100 IWG ERJ 190-100 ECJ</td>
<td>Mounting device and power supply connection for PIVOT EFB solutions onE190Rev 1; new configuration (named CONF 2) of the mechanical configuration incompatible with &quot;PIVOT&quot; cases without sliding rail and integration of anew location PWO for the USB power Rev 2; Amended inspections interval on supplement...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment 4 paragraphs: 25.856, 25.1361, 25.1309, 25.1351, 25.1353, 25.1357, 25.1360, 25.1431- CS, ...</td>
<td>- Master Data List, Rev. 10088-1-A-MDS-R04 dated 30 Apr 2024.</td>
<td>- Limitations as per §3 of Airplane Flight Manual Supplement, document referenced above. The installation of the USB power connection option is limited to the ERJ 190-100-200. Prior installation of this change/repair it must be determined that the interaction between this change/repair and any...</td>
<td>03/05/2024</td>
<td>Active</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>1007944</td>
<td>GDC ENGINEERING GmbH</td>
<td></td>
<td></td>
<td>A330-202</td>
<td>Provisional cargo compartment - A330 LEVISTh proposed design change will convert the passenger compartment into an acarcopartment.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>- Master certification document list (MCDL)</td>
<td>- The allowance to transport cargo in passenger compartments limited to 2 000 FH following installation of the change or to 31 July 2022, whichever occurs first.</td>
<td>19/07/2022</td>
<td>Active</td>
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<tr>
<td>1007949</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>FAA STC</td>
<td>ST04520CH</td>
<td>A321-253N/-271N/-272N</td>
<td>Installation of Rosemount Aerospace Aircraft Interface Device (AID) Data,Leader Provisions</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>- Rosemount Aerospace Inc. Master Data List Doc. No. D000022115, Revision A, dated 1 June 2020FAA Letter, as issued 19 July 2022Rosemount's Statement of Compliance to EASA's Airbus Certification Basis, dated 26 June 2022Certification Plan, Document No. E22011-4626-001 Revision --, dated 28 April 2022.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>26/07/2022</td>
<td>Active</td>
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<tr>
<td>1007988</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>REV. 1</td>
<td>A340-642, A340-643</td>
<td>Airbus A340 Zero Pax LOPAR</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>- Per NSI-13525-01-FMS-01 Flight Manual Supplement- EASA STC 10010624 embodied prior to 31 July 2022 is a prerequisite for the interrelationship between this change/repair and any other previously installed change/repair</td>
<td>27/03/2023</td>
<td>Active</td>
<td>1007988</td>
<td></td>
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<tr>
<td>10079801</td>
<td>SABENA TECHNICS BOC</td>
<td>REV. 1</td>
<td>A320-251N</td>
<td>Cabin reconfiguration with 186 YC seatRevision 1 of the certificate aimed to review of the certification basis.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment:</td>
<td>Certification plan reference 41052-01-A-GP-R03 dated 26 July 2022Master Data List 41052-01-MCDL-R02 dated 26 July 2022Revisions for Continued Airworthiness 41052-01-A-ICA-R00 dated 19 July 2022Later revisions of the above listed document(s) approved/ accepted by EASA.</td>
<td>As a concurrent requirement of this change, accomplishment of this STO requires the simultaneous accomplishment of the following modification:- Installation of a photo luminescent EFPN as per Luftansa Technik EASASTC No. 10014013 Rev 02.Prior to installation of this change/repair it must be determined...</td>
<td>27/07/2022</td>
<td>Active</td>
<td>10079801</td>
</tr>
<tr>
<td>10079835</td>
<td>TELEDYNE CONTROLS, LLC</td>
<td>FAA STC</td>
<td>ST05728A</td>
<td>A321-271N/-271NX/-272N</td>
<td>Installation of Teledyne Aircraft Cabin Environmental Sensor (ACES)/System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>FAA Approved Master Data List (MDL) MLD252500-A320, Rev. &quot;&quot;., dated 20 May 2021The Instructions for Continued Airworthiness are called out in the Master Drawing List (MDL) MLD252500-A320, Rev. &quot;&quot;., dated 20 May 2021Weight and Balance Manual (WBM) MLD252500-A320, dated 7 February 2022.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/07/2022</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
<th>FAA STC</th>
<th>ST 04526CH</th>
<th>MYSTERE FALCON 900 (MF900) FALCON 900EX (PF00EX)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>SKANDIA, INC</td>
<td>737-800</td>
<td>Passenger Oxygen System Replacement</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraphs were taken into account: CS-22 Amendment 3 regarding (miscellaneous Changes).</td>
<td>Master Data List (JBL - M-03736MDL-001, Issue 04, 2022-07-29-)</td>
<td>Prior to installation of this change/repair it must be determined that there is an interrelationship between this change/repair and any other previously installed change and/or repair. If no adverse effect upon the airworthiness of the aircraft is determined, the requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td></td>
<td>08/08/2022</td>
<td>Active</td>
<td>10079875</td>
<td></td>
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<tr>
<td>JET AVIATION AG</td>
<td>737-800</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of a sensor box for gliders and powered gliders with an at-tail.</td>
<td>- Anhang zum Flughandbuch - Duo Discus (T) für den Betrieb mit MASB-Messbox, Revision 1.2022-07-22-, Anhang zum Flughandbuch - Duo Discus (T) für den Betrieb mit MASB-Messbox, Revision 01, 2022-07-22-.</td>
<td>None</td>
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<td>08/08/2022</td>
<td>Active</td>
<td>10079875</td>
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<tr>
<td>AKADEMISCHE FLIEGERGRUPPE BRAUNSCHWEIG e.V.</td>
<td>737-800</td>
<td>RELOCATION, REMOVAL AND MODIFICATION OF AVIONICS AND ELECTRICAL SYSTEMS</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>Master Data List (JBL - M-03736MDL-001, Issue 04, 2022-07-29-)</td>
<td>Prior to installation of this change/repair it must be determined that there is an interrelationship between this change/repair and any other previously installed change and/or repair. If no adverse effect upon the airworthiness of the aircraft is determined, the requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td></td>
<td>08/08/2022</td>
<td>Active</td>
<td>10079875</td>
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<tr>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>737-800</td>
<td>RELOCATION, REMOVAL AND MODIFICATION OF AVIONICS AND ELECTRICAL SYSTEMS</td>
<td>The Type and OSD Certification Bases (CB) for the original product if applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>Master Data List (JBL - M-03736MDL-001, Issue 04, 2022-07-29-)</td>
<td>Prior to installation of this change/repair it must be determined that there is an interrelationship between this change/repair and any other previously installed change and/or repair. If no adverse effect upon the airworthiness of the aircraft is determined, the requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>737-800</td>
<td>RELOCATION, REMOVAL AND MODIFICATION OF AVIONICS AND ELECTRICAL SYSTEMS</td>
<td>The Type and OSD Certification Bases (CB) for the original product if applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td>Master Data List (JBL - M-03736MDL-001, Issue 04, 2022-07-29-)</td>
<td>Prior to installation of this change/repair it must be determined that there is an interrelationship between this change/repair and any other previously installed change and/or repair. If no adverse effect upon the airworthiness of the aircraft is determined, the requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.</td>
<td></td>
<td>08/08/2022</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
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<td>Status</td>
<td>System Record</td>
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<tr>
<td>HAEOD CAbIN SOUTIOnS, LLC</td>
<td>Interior Reconfiguration of a Boeing 737-800 aircraft. The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment/CS25 at amdt: 27, 25.371(a)(5), 25.371(b)(3), 25.381 (d) The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/06/2024</td>
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<tr>
<td>Q.C.M. DESIGN GmbH</td>
<td>P-RNAV (RNP-1) airworthiness approval based on existing aircraft avionics systems. The Type and OSD Certification Basis (CB) for the original product predetermines applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Maintained in accordance with L60X-QCM34-003-M1-ICA Instructions for Continued Airworthiness rev II; operated in accordance with L60X-QCM34-003-M1-AFM Aircraft Flight Manual rev 00. Duncan Aviation Master Document List 120726002 Rev A and associated FAA Form 8100-5 or later revisions of the above listed doc...</td>
<td>Limited to aircraft Learjet 60 s/n 186. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/08/2024</td>
<td>Active</td>
<td>10776561</td>
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<tr>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>Retrofit Installation of Rockwell Collins TCAS/ACAS II SW 7.0, Directional Transponders and VHF-1 Comm. The Type and OSD Certification Basis (CB) for the original product predetermines applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Installation of retrofit modification must be performed in accordance with FAEI Master Drawing and Data List designated DGL-1017. Rev. BBTG/ACAS II operation must be conducted in accordance with FAA/EVI Company Inc. Flight Manual Supplement designated FMS-20111 Rev 0 Maintenance must be conducted.</td>
<td>Limited to Aircraft Series Number 28.Bombardier TCAS II and Mode-S structural provisions. are required premature to the Collins TCAS/ACAS II and Mode-S system retrofit. If not already filled with the appropriate structural provisions, Bombardier Structural Provisions Service Bulletin designated 5Bia...</td>
<td>09/08/2024</td>
<td>Active</td>
<td>10776562</td>
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<tr>
<td>ENVOY AEROSPACE, LLC</td>
<td>Halotron BxX Fire Extinguisher Replacement/Replacement of existing Halon 1211 portable fire extinguishers with Halotron BxX portable fire extinguishers. The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment/CS25 at amdt: 25.250, 25.303(a), 25.303, 25.305(a), 25.305(a), 25.381(a), 25.601, 25.603, 25.605(a), 25.611...</td>
<td>The installation is defined with Revision A of the Envyo AerospaceMaster Data List Document E18947-2805-L001Maintenancebelievable completed in accordance with Revision A of the Envoy Aerospace Instructions for Continued Airworthiness DocumentE 18947-2805-L001Maintenancebelievable completed in accordance with Revision A of the Envoy Aerospace Instructions for Continued Airworthiness DocumentE 18947-2805-B001 or later revisions of the above listed doc...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/08/2022</td>
<td>Active</td>
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<tr>
<td>AVIONIK STRAUßBURG ENTWICKLUNGS GmbH</td>
<td>Installation of single Garmin GA 36 GSP/ WAAS Antenna on top fuselage. The Type and OSD Certification Basis (CB) for the original product predetermines applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Installed in accordance with ASR-2022-010-10-10-00 Rev 00.Maintainedin accordance with ASR-2022-010-10A-11-00 Rev 00(&gt;&gt;&lt;&lt;)= or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This installation is limited to the mechanical installation. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/08/2024</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>FLYING COLOURS CORP.</td>
<td>Installation of a Satellite TV System/intallation of equipment to support satellite TV display on cabin IFE.</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) of the later amendment: CS-22 Amd 3 paragraph(s): 22.472, 22.731, 22.732, 22.736. The requirements for environmental protection are unchanged:</td>
<td>Installed and approved in accordance with: EASA EG Dopeduring FSO-BadTölz rev 8 and Zertifizierungsplan FSO Bad Tölz dated 1/02/2021:&lt;i&gt;1&lt;/i&gt; + or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Limited to aircraft S/N 36602; airspace speed shall be kept at or below the limit as displayed in the cockpit. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>11/08/2022</td>
<td>Active</td>
<td>10079923</td>
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<tr>
<td>FLUGSPORTVERIN N BAD TOELZ e. V.</td>
<td>Increase MTOM from 770kg to 800kg</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) of the later amendment: CS-22 Amd 3 paragraph(s): 22.472, 22.731, 22.732, 22.736. The requirements for environmental protection are unchanged:</td>
<td>Installed and approved in accordance with: EASA EG Dopeduring FSO-BadTölz rev 8 and Zertifizierungsplan FSO Bad Tölz dated 1/02/2021:&lt;i&gt;1&lt;/i&gt; + or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Limited to aircraft S/N 36602; airspace speed shall be kept at or below the limit as displayed in the cockpit. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>10/08/2022</td>
<td>Active</td>
<td>10079922</td>
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<tr>
<td>COLOURS CORP.</td>
<td>Mechanical provision for the installation of an antenna, located between frame 13D and frame 13C along BL 0&lt;i&gt;&gt;&lt;/i&gt;,&lt;i&gt;&lt;&lt;/i&gt;</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) of the later amendment: CS-22 Amd 3 paragraph(s): 22.472, 22.731, 22.732, 22.736. The requirements for environmental protection are unchanged:</td>
<td>Installed and maintained in accordance with Document Number: AAFC-0102, Revision 4. Dated 4th August 2022. Operated in accordance with Document Number: AAFC-0102, Revision 3. Dated 2nd February 2022. Maintained in accordance with Document Number: APSM-0102, Revision 3. Dated 12th May 2022.&lt;i&gt;1&lt;/i&gt; + or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>10/08/2022</td>
<td>Active</td>
<td>10079921</td>
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<tr>
<td>DOA AEGISOLOBAL LIMITED</td>
<td>INSTALLATION OF FORWARD CABIN CLOSETS: This modification installs seat track mounted under seat stowage in the forward and righthand cabin area of the aircraft.</td>
<td>The change is defined with Issue 2 of AEG Master Data List Document AES-737-6530M DL. The installation shall be maintained in accordance with the following paragraph(s) of the later amendment: CS-22 Amd 3 paragraph(s): 22.472, 22.731, 22.732, 22.736. The requirements for environmental protection are unchanged:</td>
<td>Installed and maintained in accordance with Document Number: AAFC-0102, Revision 4. Dated 4th August 2022. Operated in accordance with Document Number: AAFC-0102, Revision 3. Dated 2nd February 2022. Maintained in accordance with Document Number: APSM-0102, Revision 3. Dated 12th May 2022.&lt;i&gt;1&lt;/i&gt; + or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>11/08/2022</td>
<td>Active</td>
<td>10079922</td>
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<tr>
<td>GFD GmbH</td>
<td>36A, 35A</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) of the later amendment: CS-22 Amd 3 paragraph(s): 22.472, 22.731, 22.732, 22.736. The requirements for environmental protection are unchanged:</td>
<td>Installed and maintained in accordance with Document Number: AAFC-0102, Revision 4. Dated 4th August 2022. Operated in accordance with Document Number: AAFC-0102, Revision 3. Dated 2nd February 2022. Maintained in accordance with Document Number: APSM-0102, Revision 3. Dated 12th May 2022.&lt;i&gt;1&lt;/i&gt; + or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>10/08/2022</td>
<td>Active</td>
<td>10079921</td>
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<tr>
<td>AIR PROJET</td>
<td>RF 4 RF 3 FOURNIER RF 4 D FOURNIER RF 3</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) of the later amendment: CS-22 Amd 3 paragraph(s): 22.472, 22.731, 22.732, 22.736. The requirements for environmental protection are unchanged:</td>
<td>Installed and maintained in accordance with Document Number: AAFC-0102, Revision 4. Dated 4th August 2022. Operated in accordance with Document Number: AAFC-0102, Revision 3. Dated 2nd February 2022. Maintained in accordance with Document Number: APSM-0102, Revision 3. Dated 12th May 2022.&lt;i&gt;1&lt;/i&gt; + or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>10/08/2022</td>
<td>Active</td>
<td>10079920</td>
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<tr>
<td>BERINGER Wheel and Brake upgrade for Fournier RF 3&lt;i&gt;&lt;&lt;/i&gt;4&lt;i&gt;&gt;&lt;/i&gt; (Roue et frein BERINGER Fournier RF 3&lt;i&gt;&lt;&lt;/i&gt;4&lt;i&gt;&gt;&lt;/i&gt;)</td>
<td>Mechanical provision for the installation of an antenna, located between frame 13D and frame 13C along BL 0&lt;i&gt;&gt;&lt;/i&gt;,&lt;i&gt;&lt;&lt;/i&gt;</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) of the later amendment: CS-22 Amd 3 paragraph(s): 22.472, 22.731, 22.732, 22.736. The requirements for environmental protection are unchanged:</td>
<td>Installed and maintained in accordance with Document Number: AAFC-0102, Revision 4. Dated 4th August 2022. Operated in accordance with Document Number: AAFC-0102, Revision 3. Dated 2nd February 2022. Maintained in accordance with Document Number: APSM-0102, Revision 3. Dated 12th May 2022.&lt;i&gt;1&lt;/i&gt; + or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>10/08/2022</td>
<td>Active</td>
<td>10079920</td>
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**Note:** The above information is extracted from the List of EASA Supplemental Type Certificates as of 03/07/2024.
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<th>FAA STC Number</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign N/AA</th>
<th>Models</th>
<th>Description</th>
<th>Certification Basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10079625</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>TCCA-C-LSA02-632/D, ISSUE 1</td>
<td>DHC-8-102</td>
<td>Bendix King KDM-706A GME / KFS-564A NAV</td>
<td>1 Frequency Control Retrofit</td>
<td>The Type and OSO Certification Bases (CB) for the original products/aircrafts applicable to this certificate/approval. The modifications to environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w/...</td>
<td>Installation, Removal and Modification must be in accordance with Field Aviation Master Drawing and Data List designated DDL-1088 Rev IR/Maintenance must be performed in accordance with Field Aviation Maintenance Manual Supplements designated MS/2005/5 Rev IFR later revisions of the above listed document.</td>
<td>12/06/2022</td>
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<td>10079625</td>
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<td>10079626</td>
<td>FIELD AVIATION COMPANY Inc.</td>
<td>TCCA C-LSA02-222/D, ISSUE 1</td>
<td>DHC-8-102</td>
<td>Installation of Replacement Seat Cushion/Cover Assemblies</td>
<td>The Type and OSO Certification Bases (CB) for the original products/aircrafts applicable to this certificate/approval. The modifications to environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w/...</td>
<td>Installation of Replacement Seat Cushion/Cover Assemblies to be performed in accordance with Field Aviation Company Inc. Drawings: Model 920 Seats: DWG 18322950, initial issue; Model 910 Seats: DWG 18322952, initial issue; Attendants Seats: DWG 18322953, initial issue, later revisions of the above...</td>
<td>Limited to Aircraft S/N 038. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/06/2022</td>
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<td>10079626</td>
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<tr>
<td>10079931</td>
<td>BHE &amp; ASSOCIATES, LTD.</td>
<td>FAA STC SA1296DS</td>
<td>SEE FAA AML</td>
<td>An update to the existing Collins Aerospace Integrated Flight Information System (IFIS) FSU-5010 and upgrade the existing XMWR-1000 to a XMWR-1000S (Sirus XM) in Part 23 aircraft.</td>
<td>The Certification Basis for the original products/aircrafts remains applicable to this certificate/approval. The modifications to environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List 657-00-001, Revision IR dated 27 June 2022.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/06/2022</td>
<td>Active</td>
<td>10079931</td>
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<tr>
<td>10079962</td>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>767-300</td>
<td>CABIN MODIFICATION 767-300This modification covers the removal of Passenger Seats achieving a 96C (Group 1) or a 12 BC (Group 2) configuration for supernumerary occupants.</td>
<td>The Type and OSO Certification Bases (CB) for the original products/aircrafts applicable to this certificate/approval. The modifications to environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w/...</td>
<td>The Change is defined with Issue 2 of Eirtech Aviation Services MasterDocument List Document MDU 1635-001 The aircraft shall be operated in accordance with Issue 2 of Eirtech Aviation Services Flight Manual Supplement Document FMS 1635-001 Maintenance shall be conducted in accordance with Issue 2 of E.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/06/2022</td>
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<td>10079963</td>
<td>AVADECO GmbH</td>
<td>ATR72-212A</td>
<td>Defibrillator Installation</td>
<td>The Type and OSO Certification Bases (MDL-02022) with Rev. 00 dated 26.07.2022 for later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Master Documentation List Document MDL-02022 with Rev. 00 dated 26.07.2022 for later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/06/2022</td>
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<td>10079963</td>
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<td>EASA STC no.</td>
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<td>Foreign/NAA (if rt)</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>10079967</td>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>TCCA C-LSA09-077/D, ISSUE 2</td>
<td>DHC-8-102</td>
<td>Avionos &amp; Equipment Installations</td>
<td>The Type and OSD Certification Bases (CB) for the original production variants applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w..</td>
<td>Definition of the Type Design Change in accordance with Avmax Top Documents List 116888-400, Rev. 15. Installation of the equipment and/or system is to be done in accordance with Avmax Top Documents List 116888-401, Revision 1. Maintenance is to be performed in accordance with Western Avionics Manual...</td>
<td>19/08/2022</td>
<td>Active</td>
<td>10079967</td>
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<tr>
<td>10079988</td>
<td>BOEING IRELAND LIMITED</td>
<td>787-10</td>
<td>Introduction of PEC Cabin for 777-300, 777-300ER</td>
<td>Change / Repair Approval, YLM 787-10 Introduction of Premium Economy Cabin, S1L-00196-APP. Revision 02, dated 17th August 2022.</td>
<td>The Type and OSD Certification Bases (CB) for the original production variants applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w..</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/08/2022</td>
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<td>10079988</td>
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<tr>
<td>10079971</td>
<td>APTOZ EHF</td>
<td>777-300, 777-300ER 777-200, 777-200LR</td>
<td>Zero pass LOPACabin, with no passenger seats installed.</td>
<td>Zero pass LOPACabin, with no passenger seats installed.</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: C5-MMEL Issue 02. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w..</td>
<td>AFM supplement Ref: AFM251315-1 Rev: ICA supplement Ref: ICA251315-1 Rev: AMMEL supplement Ref: MMEL251315-1 Rev: PMaster data List: MLD251315 Rev: Aor later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>22/08/2022</td>
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<td>10079973</td>
<td>AVADÉCO GmbH</td>
<td>CL-600-2024 (RJ SERIES 900)</td>
<td>Defibrillator Installation</td>
<td>Defibrillator Installation</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements Special Condition (s): CRI F-01 &quot;Non-rechargeable Lithium Battery Installations&quot;The following paragraph(s) at a later amendment: CS-25 paragraph 25.853b Amend 27TL.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/08/2022</td>
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<td>10079973</td>
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<td>10079988</td>
<td>ISRAEL AEROSPACE INDUSTRIES, LTD IAI</td>
<td>CAAI SA299</td>
<td>737-700</td>
<td>Conversion of a Boeing 737-700 PAX airplane configuration into a SpecialFreighter (SF). This conversion to a special freighter configuration among other features includes a main deck cargo compartment and associated detection system, installation of a main deck cargo door withoutaftairs.</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): D-01 &quot;Cargo compartment fire protection of essential systems&quot; issue 2, D-02 &quot;Class E cargo compartments fire protection of essential systems&quot; issue 2, D-03 &quot;Access to class...</td>
<td>The following IAI's documents: 365-00-00-M4510 Master Data List MLD revision M-365-00-00-M4505 Aircraft Flight Manual Supplement (EASA) revision New 365-00-00-M4515 Structural Repair Manual Supplement revision D-365-00-00-M4504 Maintenance Planning Document (MPD) Supplement following Document...</td>
<td>23/08/2022</td>
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<td>EASA STC no.</td>
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<td>Foreign/NAA STC Id</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10079990</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST04107NY</td>
<td>A350-941</td>
<td>Installation of a Panasonic Avionics In-Flight Entertainment System</td>
<td>The Certification Basis for the original product remains applicable to those certificates/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Delta Engineering Master Data List 0175-11438-251L, Revision A, dated 8 March 2022. The modification is defined with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>This STC is applicable to all serial numbers listed in the Master Data List.</td>
<td>23/08/2022</td>
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<tr>
<td>10079991</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>GVI (G650)</td>
<td>Cabin Interior Modification/Modification of the Cabin to generate a 12 Passenger Seating/Configuration including an upgrade of the In-Flight Entertainment System</td>
<td>The Type and OSD Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>The Change is defined with SCANDINAVIAN AVIONICS Design Certification Documents List 3000055 DOC, issue 1. The installation is defined with the SCANDINAVIAN AVIONICS Design Technical Order 3000055-T0, issue 1.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
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<td>10079992</td>
<td>DELTA AIR LINES, INC dba DELTA TECHOPS AND/OR DTO</td>
<td>FAA STC ST00014AT</td>
<td>767-300</td>
<td>Installation of forward and aft cargo compartment floor panels</td>
<td>The Certification Basis for the original product remains applicable to those certificates/approval, except where amended by additional or later amendments if indicated on FAA STC ST00014AT. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Delta Engineering Order 1-74138-3, Revision B ((&lt;,&gt;) dated 23 September 1999((&lt;,&gt;))or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
<td>Active</td>
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<td>10079994</td>
<td>DELTA AIR LINES, INC dba DELTA TECHOPS AND/OR DTO</td>
<td>FAA STC ST01357AT</td>
<td>767-300</td>
<td>Installation of Allied Signal/Bendix Weather Radar System</td>
<td>The Certification Basis for the original product remains applicable to those certificates/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/approval.</td>
<td>Delta Air Lines Engineering Report Number 1-71535-20, IR, dated 15 April 1997((&lt;,&gt;))or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/08/2022</td>
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<td>10079998</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Certification Wireless Charging Station; this modification covers the activation of Wireless Charging Systems for Portable Electronic Devices (PED) comprising one or more Wireless Charging Stations installed in the Cabin</td>
<td>The Change is defined with Revision A of AMAC Aerospace Master Document List Document MD-ALLA001-2470; Maintenance shall be conducted in accordance with Revision A of AMAC Aerospace Instruction s for Continued Airworthiness document DL-ALLA001-2470((&lt;,&gt;))or later revisions of the above listed document(s) specified.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/08/2022</td>
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<td>STC Holder</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>BROMIA AIR MAINTENANCE AB</td>
<td>Internet connection of EFbs with WiFi</td>
<td>- AF-16-003 Rev B MCA- Approval Form (Mechanical &amp; electrical install)</td>
<td>- ICA-16-16 Rev C Instruction for Continued Airworthiness - FMS-16-16 Rev B Flight Manual</td>
<td>Prior to installation of this change/repair it must be determined that the airworthiness of the product is not adversely affected.</td>
<td>24/08/2022</td>
<td>Active</td>
<td>10100000</td>
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<tr>
<td>CHIPPWA AEROSPACE, Inc.</td>
<td>Installation of a Portable Data Loader (PDL)</td>
<td>- Chippewa Aerospace Master Drawing List (MDL) No. PDL-3170-01-0001, Rev. IR, dated 30 April 2013.&lt;(</td>
<td>)&gt;or later revisions of the above listed document(s) approved/accepted.</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the airworthiness of the product is not adversely affected.</td>
<td>26/09/2022</td>
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<td>10200014</td>
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<tr>
<td>HAECCO CABIN SOLUTIONS, LLC</td>
<td>Installation of a Honeywell Mode S Enhanced Surveillance Transponder and/ or installation of an ADS-B Out System for Hudson Bay Area Operations</td>
<td>- 1. TIMCO Master Data List 03T092R007, Revision A, dated 30 April 2014.&lt;(</td>
<td>)&gt;or later revisions of the above listed document(s) approved/accepted.</td>
<td></td>
<td>No evidence of compliance with CS-ACNS provided for the ADS-B Out system installation. Prior to installation of this change/repair it must be determined that the airworthiness of the product is not adversely affected.</td>
<td>17/04/2023</td>
<td>Active</td>
<td>10200015</td>
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<td>DOA AESGLOBAL LIMITED</td>
<td>REPLACEMENT OF G1 AND G2Galley/STHS CHANGE INTRODUCES A REPLACEMENT G1 AND G2 POSITION ON THE AIRCRAFT Rev. 1 enables removal of all installation limitations.</td>
<td>- 1. TIMCO Master Data List AES-737-4660MDL Issue 02 dated Sept 2013.&lt;(</td>
<td>)&gt;or later revisions of the above listed document(s) approved/accepted.</td>
<td></td>
<td>The aircraft must be maintained in accordance with ICA supplements/revision AES-TR-3273 Issue 01. Prior to installation of this change/repair it must be determined that the airworthiness of the product is not adversely affected.</td>
<td>17/05/2024</td>
<td>Active</td>
<td>10200018</td>
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<td>DELTA AIR LINES, INC. d/b/a DELTA TECHOPS AND/OR DTO</td>
<td>Installation of AlliedSignal TCAS II</td>
<td>- MDL 34-1074&lt;(</td>
<td>)&gt;or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedural of EU-USA Bilateral Agreement.</td>
<td></td>
<td>For the purpose of compliance with IR2017/01, concurrent installation of EASA STC 100076669 (FAA STC STG0360LA) is required. Prior to installation of this change/repair it must be determined that the airworthiness of the product is not adversely affected.</td>
<td>26/09/2022</td>
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<td>A320-232</td>
<td>Structure Reinforcement Common Design for ARINC 791 Large Antenna Installation on Airbus Single Aisle</td>
<td>- Certification Programme CO177-CEP-07 dated 25.08.2022. Component User Manual CO177-CUM-01. This STC constitutes an approval for installation of EASA Supplement, dated 13 November 1988 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>FAA STC ST01757AT</td>
<td>Active</td>
<td>29/08/2022</td>
<td>10080030</td>
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<td>A320-232</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>- Special Conditions: CRI C-0303-003. The following paragraph(s) at a later amendment: CS 25 at amendment 4paragrapghs. 25.1, 25.29, 25.561, 25.611, 25.843.</td>
<td>A320-232</td>
<td>Active</td>
<td>02/09/2022</td>
<td>10080048</td>
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<td>A320-232</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List Ref. 0887-03-A-MLD-R00. Instructions for continued airworthiness, Ref. 0887-03-A-ICA-R00 (for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>- Certification Programme CO177-CEP-07 dated 25.08.2022. Component User Manual CO177-CUM-01. This STC constitutes an approval for installation only for aircraft serial numbers as stated in the Certification Program CO177-CEP. For additional aircraft serial numbers, the requirements for full compliance have to be...</td>
<td>A320-232</td>
<td>Active</td>
<td>02/09/2022</td>
<td>10080059</td>
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</table>
10080065  GDC ENGINEERING GmbH  REV. 1  A330-343 A330-202  Airbus A330 PAK to Freighter MEDUSA Conversion from a passenger aircraft configuration to an "AvenirMEDUS" freighter, an intermediate Class E freighter type with maindeck cargo door and no installation of a cargo loading system. Revision 1: adds model A330-343 to the applicability...

The Type and OS Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): E-124: Corresponds to the E-125: Class E cargo compartment fire protection of essential systems dated 27th March 2006.


Aeroplanes modified per this STC must not be equipped with stand-alone/non-integrated alerting system. Aeroplanes should not have subsequent cockpit modifications embodied without application to EASA and consultation to the STC holder. Prior to installation of this change/repair it must be done...

10080125  HONEYWELL INTERNATIONAL Inc.  FAA ST02737LA  777-300, 777-300ER, 777F, 778-200, 777-200LR  Installation of Honeywell LED Modules in the Forward Position

- Honeywell Master Drawing List 756-0280-001, revision - dated 13/11/2020
- Honeywell Master Instructions for Continued Airworthiness document number 756-0280-800, revision - dated 19/28/2020
- Or later revisions of the above listed document(s) approved/approved on behalf of EASA in accordance with the Te...n.

Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10080130  L2 CONSULTING SERVICES Inc.  FAA ST09927AC  737-900, 737-900ER, 737-900ER, 737-800, 737-8  Installation of a Gables Audio Management System K01 configuration: SATVOICE Panel

The Certification Basis for the original product remains applicable to: SATVoice certificat...approval. The requirements for environmental protection and the associated certified noise... is dated 27th March 2006.

- Master Data List 2351-B130-MLD, Revision B, dated 10 March 2022
- Configuration K01 - Airplane Flight Manual Supplement 2351-B130- AFMS-01, Revision IR, dated 30 March 2022

Conformations K01 and K02 can be installed independently or both configurations can be installed. Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10080133  AIR PROJET  Fournier RF 4 D  Wingtip Display Pod RF4Wingtip pod that carries pyrotechnic charges and lights for aerobatic displays.


Applicability limited to MSN 4014 and 4030. Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

10080135  LIUHTHADA TECHNIK AG  787-8, 787-9, 787-10  Inst. of non-electrical FPM System installation of non-electrical Floor Path Marking System: The system consists of a continuous photoluminescent strip down both sides of the cabin aisle. The photoluminescent strip will be charged by the cabinlighting and will then glow greenish at total darkness.

The Type and OS Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and or emissions levels of the product are unchanged and remain applicable to this certificate/approval.

Supplemental Type Definition Document: STOCD002-2210G Issue: 1/0(Airplane Flight Manual Supplement: A-G0-2211- AFM Issue: 1/0) Instructions for Continued Airworthiness: A-00-2211-ICAIssue 20/Master Minimum Equipment List Supplement: A-00-2211-MMEL Issue: 1/0 later revisions of the above listed documents...

Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.
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<th>EASA STC no.</th>
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<th>STC Ref.</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
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<tr>
<td>10080138</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>HT69-B</td>
<td>Cabin Configurations (Modification 2188D)</td>
<td>The Type and OSD Certification Bases for the original product are amended by the following: - additional or alternative airworthiness requirements: Equivalent Safety Finding(s): ESB -E29.807-01. Use of flight crew emergency exits for passenger evacuation with the aircraft on its side on 27th OCT 2...</td>
<td>- Master data list (mdl) ref. ae21080DML, issue 0 dated 14/09/2022 - Aircraft flight manual supplement ref. ae21080FM, issue 0 dated 14/09/2022 - Instruction for continued airworthiness ref. ae21080CA, issue 0 dated 14/09/2022 or later revisions of the above listed document(s) approved/ accepted/...</td>
<td>Prior to installation of this change/repair it must be determined that there is an interrelationship between this change/repair and any other previously installed change and/or repair it will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/09/2022</td>
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<td>10080178</td>
<td>AVCON INDUSTRIES, Inc.</td>
<td>FAA STC SA243CE</td>
<td>B300 200, 200XT, 200T, B200, B200C</td>
<td>Installation of provisions for a scanner, with optional drop hatch.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
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<td>22/09/2022</td>
<td>Active</td>
<td>10080178</td>
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<td>10080184</td>
<td>ARMSTRONG AEROSPACE Inc dba ASTRONICS ARMSTRONG AEROSPACE</td>
<td>FAA STC ST0275CH</td>
<td>A319-112</td>
<td>Installation of a Aircraft Passenger Cabin Modifications</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td></td>
<td></td>
<td>31/01/2023</td>
<td>Active</td>
<td>10080184</td>
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<td>10080191</td>
<td>BELL TEXTRON, Inc.</td>
<td>FAA STC SR0105DE</td>
<td>3.429 2.298L4, 407 2.298L6, 206L-1, 206L-3 2.298A, 206B 1.505</td>
<td>Installation of QuantFLY Data Acquisition System</td>
<td>The Type and OSD Certification Bases for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated airworthiness rules/ or emissions levels of the product are unchanged, and remain applicable to this certificate/approval with...</td>
<td></td>
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<td>02/02/2023</td>
<td>Active</td>
<td>10080191</td>
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<tr>
<td>10080201</td>
<td>STRAIGHT FIGHT, LLC</td>
<td>REV. 1 FAA STC SA00516DE</td>
<td>206H, T206H</td>
<td>Modification of aircraft floor to create 20 inch camera holeRevision 1 is an administrative re-issuance of EASA STC number 2004-3216 dated 30 March 2004.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated airworthiness rules/ or emissions levels of the product are unchanged, and remain applicable to this certificate/approval w...</td>
<td></td>
<td></td>
<td>27/02/2022</td>
<td>Active</td>
<td>10080201</td>
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### List of EASA Supplemental Type Certificates

<table>
<thead>
<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC/ID</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tbody>
<tr>
<td>10080232</td>
<td>NORTHWEST AEROSPACE TECHNOLOGIES NC.</td>
<td>FAA STC STO04699E</td>
<td>777-300ER</td>
<td></td>
<td>Interior refresh and installation of interior components</td>
<td>The Type and OISD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: CS25 amd. 0: 25.29, 25.30, 25.561, 25.601, 25.771, 25.789, 25.1431, 25.1529, 25.1541, 25.1581 CS25 at amd. 1: 25.305.</td>
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<td>19/10/2022</td>
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<td>10080235</td>
<td>ALAMO ENGINEERING GmbH</td>
<td>DORNER 328-100</td>
<td></td>
<td></td>
<td>STRUCTURAL INTEGRATION OF FLIR SYSTEM REPLACEMENT</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: JAR 25, Amendment 12: Amendme...</td>
<td>INSTRUCTIONS FOR CONTINUED AIRWORTHINESS ICA 22A-11A-01 Rev A 2022/09/30&lt;</td>
<td>Functional aspects of the FLIR system are not included in this approval.</td>
<td></td>
<td>28/06/2022</td>
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<td>10080241</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A350-941</td>
<td></td>
<td>A350 – Cabin Completion/installation of cabin interior, electrical and mechanical systems and structural attachments throughout the cabin.</td>
<td>The Type and OISD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): SC D-300, 01 Personal injury criteria orthofonic testing of sole facing seats: The requirements for environmental protection and the a...</td>
<td>Supplemental Type Design STDD00-20/02M Issue 01 Revision 01/02, Revision A 2022/09/30&lt;</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will not introduce a adverse effect upon the airworthiness of the product.</td>
<td></td>
<td>29/09/2022</td>
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<tr>
<td>10080242</td>
<td>MAGNETIC MRO AS</td>
<td>A320-214</td>
<td></td>
<td>A320 cabin reconfigurationThis STC covers the reconfiguration of the Airbus A320 Series aircraft form dual class 132 PAOLOPA to a single class 180 PAO all economy LOPAPA, part of aircraft redelivery program along with other cabin refurbishment works.</td>
<td>The Type and OISD Certification Bases for the original product as produced by the company does not apply to this certificate. The following paragraph(s) at a later amendment: CS 25 at amd. 0: 25.859(a), 25.860(a), 25.869 (4x)1(3)(b)(1). The requirements for environmental protection and the a...</td>
<td>Certification Programme ref MMRO-C-0105-CP-01 Issue 4/02, Revision A 2022/09/30&lt;</td>
<td>This STC is applicable to A320-214 M8N 4273 only. This STC can be applied on aircraft only provided that Airbus S2820-52-1162 (Type III Over Wing Ext activation) is also concurrently applied. and that it is checked prior to release to service that the airworthiness status is cleared.</td>
<td></td>
<td>29/09/2022</td>
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<td>STC holder</td>
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<td>Description</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Lease and Operations</td>
<td>Date of Issue</td>
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<tr>
<td>STERLING CORPORATION/ BIA STERLING HELICOPTER</td>
<td>FAA ST11285DS</td>
<td>An update to the existing Collins Aerospace Integrated Flight Information System (IFIS) FSU-5010 and upgrade the existing XMWR-1000 toa XMWR-1000S (Sirinx XM) in Part 25 aircraft</td>
<td>1. Master Data List, 665-00-0014 Revision A, dated 22 June 2022, 2. Instructions for Continued Airworthiness, 665-00-0014, Rev. A, dated 21 June 2022 or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation/Procedures of...</td>
<td>This STC is valid for aircraft types and models approved by EASA and listed on the FAA STC Document.</td>
<td>03/09/2022</td>
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<td>10080290</td>
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<tr>
<td>HELICOPTER ASSOCIATES, LTD.</td>
<td>FAA ST11258DS</td>
<td>Installation of an LS400NR-13,8400T LED sterile interior light assembly.Rev. 1 corrects the company name.</td>
<td>- MDL Document Number 910109, Rev. A, dated February 25, 2022, ICA Document Number 910110, Rev. C, dated February 25, 2022 3. SABENA BHE &amp; ASSOCIATED TECHNICAL</td>
<td>The Bell 212 or 412 series helicopter must be OEM equipped with a PAR6 searchlight at the approximate OEM location on the belly of the aircraft. Prior to installation of this change/repair it must be determined that the installation's relationship between this change/repair and any other previously installed change/repair...</td>
<td>03/10/2022</td>
<td>Active</td>
<td>10080310</td>
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<tr>
<td>EBRA SABENA TECHNICS BC</td>
<td>A321-232</td>
<td>The Certification Basis for the original product remains applicable to the following additional or alternative environmental requirements: CS-29 Amendment 8 for parts changed or affected by the change. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>- FAA STC ST11285DS. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>The Certification Basis for the original product remains applicable to the following additional or alternative environmental requirements: CS-29 Amendment 8 for parts changed or affected by the change. The requirements for environmental protection and the associated certified noise and/or emissions levels of the or...</td>
<td>04/10/2022</td>
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<tr>
<td>ETHIO AIRWAYS ENGINEERING L.L.C.</td>
<td>A320-251N</td>
<td>Cabin Reconfiguration from 195 Pax (195 Y/C) to 220 Pax (220 Y/C).</td>
<td>Master Data List, Ref. 40955-01-A-ML-DL01 dated 30th September 2022, 2. Instructions for Continued Airworthiness, Ref. 40955-01-A-ML-DL01 dated 30th September 2022 or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation/Procedures of...</td>
<td>As a concurrent requirement of this change, accomplishment of this STC requires the simultaneous accomplishment of the following modification: Installation of a photo-luminescent EPFM as per Lufthansa Technik's Engineering Bulletin, approved by EASA STC, 10014013 Rev. 3. The EPFM STC/Ref. EASA 10014013 R...</td>
<td>04/10/2022</td>
<td>Active</td>
<td>10080365</td>
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<tr>
<td>ARCHEION HOLDINGS, L.L.C.</td>
<td>FAA STC SA09887AC</td>
<td>The Certification Basis for the original product remains applicable to the following additional or alternative environmental requirements: The Type and OSD Certification Bases for the original product as amended by the following additional or alternative environmental requirements: The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List, 221250-MDL, Revision F, dated 4 August 2022. 2. Archeion Holdings Instructions for Continued Airworthiness, 221250-ICA Revision R, dated 7 June 2022 or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation/Procedures of...</td>
<td>Prior to installation of this change/repair it must be determined whether the installation's relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product</td>
<td>05/10/2022</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>AIRBUS TRANSPORT INTERNATIONAL INC</td>
<td>A300F4-605ST</td>
<td>Special installation of non-active sat/beam 1 of AT1 modification ST25MOD0020 (applicant’s reference)</td>
<td>- Change Approval Sheet (CAS) ST25MOD0020E657 issue 1 dated 05 October 2022. - Change Certification Plan (CCP) ST25MOD0020E657 issue 1 - Non active sat/beam in CTH44 antenna + Heater/Transport dated 22 September 2022: Approved Document Dossier (ADD) ST25MOD0016E7 ADD issue 2 including AFM, WBM and MWE.</td>
<td></td>
<td>The transported satellite is expected to contain active systems which have not been assessed as part of this change. Themanned flights are therefore expected to be subject to an airworthiness exemption granted by the competent authority. Prior to installation of this change/repair it must be determined.</td>
<td>05/10/2022</td>
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<td>10082072</td>
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<tr>
<td>LEARJET Inc.</td>
<td>ST00763W</td>
<td>Installation of Dual Rockwell Collins RTU-4220 Radio Tuning Units (RTU) System.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>(1) Learjet Master Document List M100264CT, Revision D, dated 9 June 1999(+-) (2) Learjet, Inc. Document No. WW741, Airplane Flight Manual Supplement, dated 17 June 1995(+-) (3) Prior revisions of the above listed document(s) approved/ accepted on behalf of EASA in accordance with the Technical Implement...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/10/2022</td>
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<td>10082077</td>
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<tr>
<td>L2 CONSULTING SERVICES Inc.</td>
<td>ST111265C</td>
<td>Installation of the Indium Communications System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 amdt 0: 25.26(a), 25.365(a)(11)(4), 25.56(1)(c), 25.57(1a)(b), 25.60(3), 25.62(5)(a), 25.79(9)(a), 25.86(9)(a) (1), 25.98(1a)(3)(c), 25.135(1a), 25.135(9)(a), 25.135(a) 2.</td>
<td>Master Drawing List 2325- B78-MDL-16-Revision A, dated 19 July 2013(+-) &gt;Airplane Flight Manual Supplement 2325-878-01AFM, Revision R, dated 13 November 2013(+-) &gt; Prior revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implement...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/11/2022</td>
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<td>DELTA AIR LINES, INC./DELTA TECHOPS AND/OR DTO</td>
<td>ST03059D</td>
<td>Installation of a Delta modified forward trunnion bearing housing and a Delta manufactured bearing retainer bolt to allow forced lubrication of the respective joints</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
<td>Data E. O. 1-16544-10, dated 8 August 1992(+-) Prior revisions of the above listed document(s) approved/ accepted on behalf of EASA in accordance with the Technical Implement Procedures of EU USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/10/2022</td>
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<td>10080319</td>
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<tr>
<td>DELTA ENGINEERING CORPORATION</td>
<td>ST034124Y</td>
<td>Installation of a Viavi Satcom Antenna and relocation of VHF 3 antennas (as required) per following configurations: Configuration 1: Mantarray-Configuration 2: Demot-Configuration 3: Thales to Kulanay-Configuration 4: Mantarray to Kullanay-Configuration 5: Thales to Mantarray-Configuration...</td>
<td>- No antennas of any size may be installed within 36” FWD or AFT of there without additional flight testing for vibration and buffeting. Configuration 3 and 5 of the Master Data List requires prior installation and validation by EASA of FAA STC ST032518CH Prior to installation of this change/repair...</td>
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<td>10/10/2022</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>TELAIR US LLC</td>
<td>Installation of the Main Deck Cargo Loading System</td>
<td>STC 62-004, Drawing List for 69000- Series Cargo Handling System.U.S. Cargo Systems Report Number RPT-3880, Revision 3.0, dated 10/06/2019 Higher, 5-6070-1 BCF PIN</td>
<td>69000-Series MDCLS, Weight &amp; Balance Manual Supplement,U.S. Cargo Systems Report Number RPT-3919, Revision 1.0, dated 15/September 2...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/10/2022</td>
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<tr>
<td>DIAMOND AIRCRAFT INDUSTRIES GmbH</td>
<td>Business Seating Configuration 1st Passenger Row/This STC provides an alternative configuration for the DA 62 w/ more comfort in the 1st passenger row. The 1st passenger row is limited to 2 passengers with installation of an armrest on the middle seat. The DA 62/1st passenger row 2/3 backrest is reple...</td>
<td>The Type and OSD Certification Basis (CB) for the original product/producing companies applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval w...</td>
<td>Master Document List AES-190-0546 MDL issue 1AFMS No 99 AFMS Supplement Issue 1COMV No 87 Cabin Crew Operating Operation Issue 1 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/10/2022</td>
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<td>DOA AESGLOBAL LIMITED</td>
<td>ERJ 190-100 STD/ERJ 190-100 SR/ERJ 190-100 LR/ERJ 190-100 SW/ERJ 190-100 ECJ</td>
<td>The Type and OSD Certification Basis (CB) for the original product/producing companies applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval w...</td>
<td>Certification Program CFP-365-508-23-01, Revision 0.3, dated 17/10/2022</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/10/2022</td>
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<tr>
<td>SPAES GmbH &amp; Co. KG</td>
<td>AS 365 N2, AS 365 N3</td>
<td>The Type and OSD Certification Basis (CB) for the original product/producing companies applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval w...</td>
<td>Installed in accordance with Installation instruction document ref:617-10030-ICN-0001 rev. A. Maintained in accordance with ICA Supplement document ref. 617-10030-ICN-0001 rev. B. Other revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Limited to aircraft s/n 532 PRIOR to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/10/2022</td>
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<tr>
<td>MONTI CARLO HELICOPTERE SERVICE(MHS)</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>The Type and OSD Certification Basis (CB) for the original product/producing companies applicable to this certificate/ approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the product are unchanged remain applicable to this certificate/ approval w...</td>
<td>Installed in accordance with Installation instruction document ref:617-10030-ICN-0001 rev. A. Maintained in accordance with ICA Supplement document ref. 617-10030-ICN-0001 rev. B. Other revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/01/2023</td>
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<td>EASA STC No.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10080448</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV-1</td>
<td>FAA ST0430AT-D</td>
<td>GULFSTREAM G150</td>
<td>Installation of a Future Air Navigation System (FANS) 1/A with datalink recording system</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): FAA SC No. 25-661-25C- ‘‘Non-rechargeable Lithium/Battery Installations’’ The requirements for environmental protection and the associate…</td>
<td>Index DTL CE00218MR0001 Revision -‘-</td>
<td>dated October 14, 14/10/2016Flight Manual Supplement CEC140988M001 Revision ‘‘-’’ dated October 14, 14/10/2016</td>
<td>The following STCs are required as part of this STC/STC ST0378AT-D: Installation of Rockwell Collins Flight Management System (FMS) Upgrade providing Wide Area Augmentation System (WAAS) with自治区 Precision with Vertical Guidance (LPV) capabilities/STC ST10132G-D Installation of a VHF COMM 3 An…</td>
<td>14/10/2022</td>
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<td>10080451</td>
<td>OGMA INDUSTRIA AERONAUTICA DEPORTUAL, S.A.</td>
<td>REV. 1</td>
<td>EMB-135BJ</td>
<td></td>
<td>Installation of the Utility Installations, ID active</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): FAA SC No. 25-661-25C- ‘‘Non-rechargeable Lithium/Battery Installations’’ The requirements for environmental protection and the associate…</td>
<td>AFM supplement ref: 18P12MS0003 Initial Issue dated 24th October 2022ICA supplement ref: 18P12MS0002 Initial Issue dated 24th October 2022</td>
<td>The following STCs are required as part of this STC/STC ST0378AT-D: Installation of Rockwell Collins Flight Management System (FMS) Upgrade providing Wide Area Augmentation System (WAAS) with自治区 Precision with Vertical Guidance (LPV) capabilities/STC ST10132G-D Installation of a VHF COMM 3 An…</td>
<td>26/10/2022</td>
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<td>10080452</td>
<td>AVMAX AVIATION SERVICES Inc.</td>
<td>DHC-8-315 DHC-8-314 DHC-8-311 DHC-8-301 DHC-8-202 DHC-8-201 DHC-8-106 DHC-8-103 DHC-8-102</td>
<td></td>
<td>Universal Avionics EFI-880R MFD Installation</td>
<td>The Type and OSD Certification Bases (CB) for the original productremains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval w…</td>
<td>AFM supplement ref: 18P12MS0003 Initial Issue dated 24th October 2022ICA supplement ref: 18P12MS0002 Initial Issue dated 24th October 2022</td>
<td>Aeroplanes modified by this STC must be operated in accordance with AFMsupplement ref: 18P12MS0003, and maintained in accordance with ICA supplement ref: 18P12MS0002</td>
<td>Prior to installation of this STC/STC ST0378AT-D: Installation of Rockwell Collins Flight Management System (FMS) Upgrade providing Wide Area Augmentation System (WAAS) with自治区 Precision with Vertical Guidance (LPV) capabilities/STC ST10132G-D Installation of a VHF COMM 3 An…</td>
<td>26/10/2022</td>
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<tr>
<td>10080460</td>
<td>AGUSTAWESTLIA N D PHILADELPHIA PC RPORTION</td>
<td>FAA STC SR04231NY</td>
<td>AW119MKII</td>
<td>Installation of the Utility Installations, ID No. AWPC-ENG-17-037-025/Exterior Cabin Boarding Lights* and G24-6 <em>Auxiliary Provisions /Relocations: Circuit Breaker Terminal Block Rail</em>.</td>
<td>The Certification Basis for the original productremains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
<td>AWPC-ENG-17-033 Rev-‘- AW119 MKII Certification Plan for Law Enforcement Utility Helicopters/AWPC- ENG-20-052 Rev A, AW119MKII Certification Plan for Amendment 1 toAWPC STC SR04233NY/AWPC Installation Master Data List Report No. AWPC- ENG-17-037 Rev E dated 16 April 2021</td>
<td>Prior to installation of this STC/STC ST0378AT-D: Installation of Rockwell Collins Flight Management System (FMS) Upgrade providing Wide Area Augmentation System (WAAS) with自治区 Precision with Vertical Guidance (LPV) capabilities/STC ST10132G-D Installation of a VHF COMM 3 An…</td>
<td>26/10/2022</td>
<td>Active</td>
<td>10080460</td>
<td></td>
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<tr>
<td>10080470</td>
<td>AVIATION TRADERS LIMITEDEUROPE S.A.S.</td>
<td>A300 F4-203 A300 B4-220, A300 C4-203 A300 B4-203 A300 B4-120 A300 B4-103 A300 B2-30, A300 B4-2C A300 B2-202, A300 B2-203 A300 B2-1C, A300 B2-1A</td>
<td>A300 B1, A300 B2-1A</td>
<td>Installation of ADS-B Out Functionality/This change aims at installing an ADS-B Out capability on Airbus A300 models.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): FAA SC No. 25-661-25C- ‘‘Non-rechargeable Lithium/Battery Installations’’ The requirements for environmental protection and the associate…</td>
<td>- Certification Programme - CP-13685-0001 Issue 5 dated 14/10/2020 Master Document List MDL-13685-0001 Issue 4 dated 14/10/2022 Aircraft Flight Manual Supplement - 13685-AFM-0001 Issue 2 dated 14/10/2022</td>
<td>Prior to installation of this STC/STC ST0378AT-D: Installation of Rockwell Collins Flight Management System (FMS) Upgrade providing Wide Area Augmentation System (WAAS) with自治区 Precision with Vertical Guidance (LPV) capabilities/STC ST10132G-D Installation of a VHF COMM 3 An…</td>
<td>27/10/2022</td>
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List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<th>Certification basis</th>
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<th>Limitations/Conditions</th>
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<th>System Record</th>
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<tbody>
<tr>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>10080482</td>
<td>GIVH-12528-01-MOD-01 - 189Y PAX LOPATs modification introduces a cabin layout using 63 economy class single seats generating a 189 passengers seating configuration.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: JAR 25.692 (c) (5) and (6)(g) at JAR 25 Change 13 The requirements for environmental protection and the associated...</td>
<td>GIVH-12528-01-MOD-01</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/10/2022</td>
<td>Active</td>
<td>10080482</td>
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<tr>
<td>MICRO AERODYNAMICS, INC.</td>
<td>10080486</td>
<td>FAA STC SA5789NM</td>
<td>Installation of vortex generators on the wing and vertical tail andstrakes on the engine nacelles.</td>
<td>The Certification Basis for the original product remains applicable to this certification approval, except where amended by the following amendments if indicated on FAA STC SA5789NM. The requirements for environmental protection and the associated certified noise and/or emissions levels of the...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/11/2022</td>
<td>Active</td>
<td>10080486</td>
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<tr>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>10080502</td>
<td>Airbus A320 180 Pax LOPA Installation/Installation of new passenger seats (Recaro 3530A Seats) are installed to accommodate a 180 passengers seat installation in a single-class configuration.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: JAR 25 Change 13: 25.692 (a) (b) (c) 25.812 (e) 25.853 (b) (c) 25.26 issue 03 paragraphs: 25.60 (b) 25.66 (a) ...</td>
<td>ICA supplement: GIVH-13104-02-ICA-01 issue 01 dated 05 Sept 2022/MDL: GIVH-13104-02-MDL-01 issue 01 dated 18 Oct 2022or later revisions of the above listed document(s) approved/ acceptable under the EASA system.</td>
<td>The aircraft must be maintained in accordance with ICA supplement referenced GIVH-13104-02-ICA-01 Issue 01. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/11/2022</td>
<td>Active</td>
<td>10080502</td>
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<tr>
<td>E A D AEROSPACE AIRWORTHINESS TAI/TECHNICAL</td>
<td>10080508</td>
<td>Airbus A320-235N-212/220/235I/251N/252N/253I-212/214/215/-216</td>
<td>Installation of new passenger seats (Recaro 3530A Seats) are installed to accommodate a 180 passengers seat installation in a single-class configuration.</td>
<td>Master Draw list: 01577-EAS-A-MDL Rev C dated 14 October 2022 ICA supplement: 01577-EAS-A-ICA-01 Rev A dated 22 July 2022or later revisions of the above listed document(s) approved/ acceptable under the EASA system.</td>
<td>Aircrafts modified by this STC must be maintained in accordance with ICA supplement referenced 01577-EAS-A-ICA-01. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/11/2022</td>
<td>Active</td>
<td>10080508</td>
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<tr>
<td>PMV ENGINEERING</td>
<td>10080546</td>
<td>CAB-04760 - 189 Y/C seats installation</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: CS 22 Amendment 2, See Certification Program. The requirements for environmental protection and the associated...</td>
<td>Master Document List: CAB-04760-MDL-Rev 00 Instructions for continued airworthiness CAB-04760-ICA Rev 00 or later revisions of the above listed document(s) approved/ acceptable under the EASA system.</td>
<td>Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change/repair and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/11/2022</td>
<td>Active</td>
<td>10080546</td>
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<td>M&amp;D FLUGZEUGBAU GmbH &amp; Co. KG</td>
<td>10080554</td>
<td>LS 4, LS 4-4, LS 4-4B</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: CS 22 Amendment 2, See Certification Program. The requirements for environmental protection and the associated...</td>
<td>- LS 4 Jet Sustainer Flight Manual Supplement, Issue: 02, Date: 07.07.2022; LS 4 Jet Sustainer Maintenance Manual, Issue: 02, Dated 31.08.2022; Master Document List, Rev. 01, Dated 22.08.2022 or later revisions of the above listed document(s) approved/ acceptable under the EASA system.</td>
<td>None</td>
<td>08/11/2022</td>
<td>Active</td>
<td>10080554</td>
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<td>STC No.</td>
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<td>Rev. Level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10080555</td>
<td>DOGA21 LIM E DIT</td>
<td>A320-251N</td>
<td></td>
<td></td>
<td>Installation of a VIP Cabin Interior</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements. Special Condition(s): F-115 “Oxygen requirement qualification above 40,000” Issue 2. The requirements for environmental protection and the associated certified...</td>
<td>Certification Programme SYM1-0000RC000 rev AAMF Supplement SYM1-0000RM500 rev AMMEL Supplement SYM1-0000RM19 Initial issue/OSD CCD Aircraft difference table SYM1-0000RC004 rev Aqr later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>08/11/2022</td>
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<td>10080564</td>
<td>AVIONITEC Ltd.</td>
<td>340</td>
<td></td>
<td></td>
<td>Installation Sky 497 TAS System including TAP Antenna. Exchange WX Radar by a Garmin GOMM 75. Exchange of the L-5 Domescope System WX-1060 by an Exastant WX-500 System</td>
<td>The Type and OSD Certification Bases (OB) for the original product as applicable to this certificate approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval w...</td>
<td>Installed in accordance with Master Document List No. 13161-17 dated 16/8/2022. Installed in accordance with Instruction for Continuing Airworthiness 13161-015 dated 16/8/2022 and Static Report 1PB_HB-LSR80083112 dated 25/8/2022. operated in accordance with AIRCRAFT FLIGHTMANUAL SUPPLEMENT No. 13161-01...</td>
<td>Limited to aircraft S/N 340-0180 Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
</tr>
<tr>
<td>10080565</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>560XL</td>
<td></td>
<td></td>
<td>Installation of Garmin G5000 Integrated Flight Deck</td>
<td>The Type and OSD Certification Bases (OB) for the original product as applicable to this certificate approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval w...</td>
<td>Master Drawing List (MDL), Garmin International, Inc. document N.005-01202-00, Revision 20&lt;,&gt; dated 15th September 2022. Airplane Flight Manual Supplement, Garmin International, Inc. document N.005-02313-03, Revision 4&lt;,&gt; dated 15th September 2022&lt;,&gt;</td>
<td>Eligible aircraft are serial numbers 560-5501 through 560-5510 (Excelent serial numbers 560-5501 through 560-6000). FS510 EASA-Certified Configuration must be loaded to the G5000 avionics&lt;,&gt;</td>
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<td>10080607</td>
<td>ROSEN SUNVISOR SYSTEMS LLC</td>
<td>60, 60, 60</td>
<td></td>
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<td>Cockpit Sun Visor installation</td>
<td>The Certification Basis for the original product remains applicable to this certificate approval except where amended by additional or later amendments if indicated on FAA STC SA3597NM. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval w...</td>
<td>MDL 9050-0116-001 Rev U, Number RBB-000L, dated 21 September 2022&lt;,&gt;</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10080616</td>
<td>AERO SEKUR S.P.A</td>
<td>AW169, AW189</td>
<td>AW139, AW139</td>
<td></td>
<td>The scope of this change is to introduce a new pyrotechnical actuator assembly kit PN 318783A as a replacement for the 0000RM500 now installed on the AW139/ AW139, AW169 and AW189 helicopters models</td>
<td>The Type and OSD Certification Bases (OB) for the original product as applicable to this certificate approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval w...</td>
<td>Master Drawing List Document MDW 318783A Rev B, Master Document List M3D, 318783A Rev.F, Master Document List M3D, 318783A Rev.C, Master Document List M3D, 318783A Rev.C.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10080626</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>MEDICAL OXYGEN BOTTLE INSTALLATION</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Equivalent Safety Findings.</td>
<td>F-03/0077948</td>
<td>Maximum Oxygen Working Pressure and Pressure Limiting Performance.</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>STC Twenty One Ltd Flight Manual Supplement No. S21:FM6-018 Issue 1 dated 04 November 2021</td>
</tr>
<tr>
<td>10080636</td>
<td>DOA21 LIMITED</td>
<td>REV. 1</td>
<td>A320-251N</td>
<td>Introduction of Cabin VIP Interiors. Rev. 1 Transfer of STC holder role to DOA21 Ltd.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements.</td>
<td>STC Twenty One Ltd Flight Manual Supplement No. S21:FM6-018 Issue 1 dated 04 November 2021</td>
<td>AEROTEC CONCEPT modification dossier no. AE20196 is a pre-requisite for the installation. Installation is limited to configurations as per Assessment Drawings: 2107301-0000; 2107302-0000; 2107304-0000; 2107307-0000; 2107308-0000; 2107309-0000; 2107310-0000; 2107311-0000; 2107314-0000. Subsequently...</td>
<td>24/10/2023</td>
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<td>10080682</td>
<td>SABENA TECHNICS BOD</td>
<td>BD-700-2A12</td>
<td>MISSILE AVOIDANCE SYSTEM MISSILE/MECHANICAL AND ELECTRICAL PROVISIONS FOR MISSILE AVOIDANCE SYSTEM MISSILE INSTALLATION ON BOMBARDIER G7500</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following:</td>
<td>STC Twenty One Ltd Flight Manual Supplement No. S21:FM6-018 Issue 1 dated 04 November 2021</td>
<td>Aircraft modified in accordance with this STC are considered to be an &quot;variant&quot; to the A320 passenger configurations. Cabin crew operating procedures modified in accordance with this STC must complete aircraft reference training in accordance with ORD.CC.130, which must include the approved OSD-CabinP...</td>
<td>Operations with Lasers ON are not approved by EASA. Lasers must be permanently disabled when operating under EASA rules. Prior to installation of this change/repair it must be determined that the change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/11/2022</td>
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<tr>
<td>10080707</td>
<td>SKY SERVICE F.B.O. INC.</td>
<td>REV. 1</td>
<td>TCCA SA22-59 ISSUE 1</td>
<td>Plane Simple Ku Band Satcom/Satcom Ku Band standalone installation providing high speed Ethernet and WiFi. Rev. 1 adds the TCCA STC reference.</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>Aircraft modified in accordance with this STC are considered to be an &quot;variant&quot; to the A320 passenger configurations. Cabin crew operating procedures modified in accordance with this STC must complete aircraft reference training in accordance with ORD.CC.130, which must include the approved OSD-CabinP...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/11/2022</td>
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<tr>
<td>10080708</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>REV. 1</td>
<td>DORNIER 228-200</td>
<td>ADS-B OUT on DORNIER 228</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>Aircraft modified in accordance with this STC are considered to be an &quot;variant&quot; to the A320 passenger configurations. Cabin crew operating procedures modified in accordance with this STC must complete aircraft reference training in accordance with ORD.CC.130, which must include the approved OSD-CabinP...</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/11/2022</td>
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**List of EASA Supplemental Type Certificates**

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<tr>
<td>10080718</td>
<td>ADAC HELISERVICE GMBH</td>
<td>10080739</td>
<td>EC135 T3</td>
<td>Installation of retractable LED search &amp; landing light Luminator LS400GR(LH and RH) in replacement of pre-installed Collins retractable LEDSearch &amp; landing light, as well as associated Night Vision Imaging System (NVIS) approval</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-27 Amendment 9 for the following paragraph(s): 27.29, 27.251, 27.301, 27.305(a)(27.321, 27.603; 27.60…</td>
<td>- Rotorcraft Flight Manual Supplement EC135-11951-025MFS Revision 1- Master Document List EC135-11951-001-LMD Revision 1- Maintenance Manual Supplement EC135-11951-023MMS Revision 1- Compliance Summary EC135-11951-044-RCS Revision 01 or later revisions of the above listed document(s) approved/accepted.</td>
<td>Limited to helicopters with Helionix cockpit variant. As prerequisite (pre-modification configuration) the Rotorcraft must be equipped with one or two of the following retractable LED Search &amp; landing (LS/L) lights: Collins LED SL light P/N 2LAA45436.0 (i.e. Airbus IPC 33-43-03-06- RH side). Collins…</td>
<td>24/11/2022</td>
<td>Active</td>
<td>10080718</td>
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<tr>
<td>10080731</td>
<td>STERLING CORPORATION/ BA STERLING HELICOPTER</td>
<td>FAA</td>
<td>429</td>
<td>Installation of Luminator Aerospace LS400GR LS400NR-GGR LED steerable search light assembly replacement. Rev. 1 corrects the company name.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-27 specifications at Amendment 1.27.251, 27.301, 27.303, 27.305(a)(b)(c), 27.307(a), 27.341, 27…</td>
<td>Master Drawing List, Document No. 910106, Rev. A. dated January 10, 2022. Instructions for Continued Airworthiness, Document No. 910107, Rev. C, dated February 25, 2022. Rotorcraft Flight Manual Supplement, Document No. 910146, Rev. IR, FAAApproved dated March 10, 2022.</td>
<td>The Bell 429 helicopter must be OEM equipped with a PAR 48 searchlight that the approximate OEM location on the belly of the rotorcraft. The Bell 429 helicopter must have the OEM dustcover installed. Prior to installation of this change/repair it must be determined that the interrelationship between this…</td>
<td>28/11/2022</td>
<td>Active</td>
<td>10080731</td>
<td></td>
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<tr>
<td>10080735</td>
<td>POLYTECH AERO SUPPORT INC.</td>
<td>FAA</td>
<td>737-800</td>
<td>Cabin Layout Reconfiguration change/Replacing seats: - Configuration 1. B/E Aerospace Spectrum model 1005679; seats replaced with ZIM-1005679 seats. The number of passengers is not affected and remains with 19.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: JAR 25.962 at Chang</td>
<td>Implantation shall be performed in accordance with PolytechAero Support. Modification Summary M55-14-002. Revision 10. Maintenance shall be performed in accordance with Polytech Aero Support. Instructions for Continued Airworthiness ICA-14-002. Revision for later revisions of the above listed do…</td>
<td>This STC is limited to the following MSN: Configuration 1-260689, 26069, 30069, 30705, Configuration 2: 30631. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce…</td>
<td>25/11/2022</td>
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<tr>
<td>10080737</td>
<td>REDAERO</td>
<td>FAA</td>
<td>A330-202</td>
<td>Recaro project RCP204: Cabin adaptation/Change of the seating capacity. from 314 seats to 288 seats, with 19 B/C seats and 269 Y/C seats. Installation of several equipment’s.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.854 at Amendment 1: CS 25.653 &amp; CS 25.854 at Amendment 0; CS 25.854 at Amendment 0 and CS 25.571 at Amendment 19. The requirement…</td>
<td>- Master Data List RCP2042002-MRE01 Issue 1 signed on 24 Nov 2022- Compliance Summary RCP2042001 Issue 1 signed on 24 Nov 2022 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/11/2022</td>
<td>Active</td>
<td>10080737</td>
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<tr>
<td>10080739</td>
<td>SIRIUM AEROTECH S.L.</td>
<td>FAA</td>
<td>SS50 560</td>
<td>Cesare Citation ADS-B Upgradation of ADS-B DO-260B capabilities with existing UNS-I/1x FMS andGNSS source.</td>
<td>The Type and OSD Certification Bases (CS) for the original product. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>MASTER DATA LIST SA19-245-MLD Rev. 01, AFM SUPPLEMENT SA19-245-AMFS Rev. 0/ MMEL SUPPLEMENT SA19-245-MMELS Rev. 00</td>
<td>Aircraft must be equipped with Transponder MTS-708 (P/N 9000700-5501 or 9000700-5701) and FMS 3UNI-1/Lw (P/N: 3118-30-1116) with OSC1001/1101/1.1 or higher or OSC 1000/81100.0 or higher with ADS-B FalseEnunciation provisions installed. Prior installation of this change/repair must be determined…</td>
<td>25/11/2022</td>
<td>Active</td>
<td>10080739</td>
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<td>STC Holder</td>
<td>FAA STC</td>
<td>STC ref.</td>
<td>Models</td>
<td>Description</td>
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</tr>
<tr>
<td>TECHNICS BGC</td>
<td>FAA STC</td>
<td>SR01082LA</td>
<td>212, 412, 412EP</td>
<td>Installation of a medical bariatric system</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Drawing List (MDL) EGR-655-02, Revision AB, dated August 8, 2007; (EGR-655-02) or later revisions of the above listed document(s) approved/ accepted on behalf of EASA in accordance with the Technical L...</td>
<td>Prior to installation of this change/repair it must be determined whether the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/11/2022</td>
<td>Active</td>
<td>10080751</td>
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<tr>
<td>LIFEPORT, LLC</td>
<td>FAA STC</td>
<td>SR00871LA</td>
<td>212, 412, 412EP</td>
<td>Installation of the Modular Sealing System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Drawing List for the Modular Sealing System Installation EGR-606-00, Revision 8, dated August 8, 2007; (EGR-606-00) or later revisions of the above listed document(s) approved/ accepted on behalf of EASA in accordance with the Technical L...</td>
<td>Prior to installation of this change/repair it must be determined whether the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/11/2022</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>FAA STC</td>
<td>SR02465LA</td>
<td>A321-271NX</td>
<td>Cabin Reconfiguration to 239 Pax</td>
<td>The Certification Plan - 41057-01-A-CP-W02 (Master Data List (MDL) A01) ref. 41057-01-A-CP W02 MDL Instructions for Continued Airworthiness (ICA) (C1507-01-A-ICA-R900 or later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>As a concurrent requirement of this change, the installation of the (photo-luminescent) FPM is needed. Some drawings may be defined to assist the installation which shall be done according to Engineering Bulletins/Issued by Luftansa Technik ref A320-933-0384 Rev5 and covered through STC ref 10014111 Rev 02...</td>
<td>13/12/2022</td>
<td>Active</td>
<td>10080765</td>
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<tr>
<td>AEROSERVIV, S.R.O.</td>
<td>STC Holder</td>
<td></td>
<td>VIPER SD-4 RTC</td>
<td>Installation of FLASH-R propeller on Viper SD-4 RTC and Viper SD-4 Night/FR exhaust modification.</td>
<td>The Certification Plan - 41057-01-A-CP-W02 (Master Data List (MDL) A01) ref. 41057-01-A-CP W02 MDL Instructions for Continued Airworthiness (ICA) (C1507-01-A-ICA-R900 or later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>As a concurrent requirement of this change, the installation of the (photo-luminescent) FPM is needed. Some drawings may be defined to assist the installation which shall be done according to Engineering Bulletins/Issued by Luftansa Technik ref A320-933-0384 Rev5 and covered through STC ref 10014111 Rev 02...</td>
<td>13/12/2022</td>
<td>Active</td>
<td>10080767</td>
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Note: The table above contains a list of EASA Supplemental Type Certificates with their respective details, including models, descriptions, certification basis, limitations/conditions, and dates of issue. The text is primarily in English, with some technical terms and acronyms related to aviation and aircraft modifications.
<table>
<thead>
<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Revision level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10080797</td>
<td>DRABPOL SP.</td>
<td></td>
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<td>A321-271N/-271NX/-272N</td>
<td>Installation of vaccine deployment system</td>
<td>Type and OSD Certification Bases (CB) for the original product by associated STC approval</td>
<td></td>
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<td>02/12/2022</td>
<td>Active</td>
<td>10080797</td>
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<tr>
<td>10080803</td>
<td>UMLAUT ENGINEERING GMBH</td>
<td>REV. 1</td>
<td>A321-271N/-271NX/-272N -213, A320-253N/-251NX/-272N/-231/-211,-212/-251N/-252N/-212/-214/-215/-216/-171N/-131/-132/-133/-151N/-152N/-111/-112/-113/-114-115</td>
<td>CCS+ Center Stowage Integration/Installation of an additional overhead stowage in the center aisle off the A320 family. Rev. 1 corrects the model list.</td>
<td>Type and OSD Certification Bases (CB) for the original product as amended by other previously installed or repair.</td>
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<td>06/12/2022</td>
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<tr>
<td>10080806</td>
<td>MECAER AVIATION GROUP S.r.l.</td>
<td>REV. 2</td>
<td>H160-B</td>
<td>VIP INTERIOR/Rev. 1 introduces new cabin layouts with modular 4-seater benches including configurations 6A6H0000F10401005, 6A6H0000F10402007, 6A6H0000F10401040 21002 and 6A6H0000F10401006. Rev. 2 introduces new cabin layouts with forward-facing armchairs including configurations 6A6H0000F103001003 and 6A...</td>
<td>The Type and OSD Certification Bases (CB) for the original product by associated STC approval.</td>
<td>Certification programme No. 6A6H0000F177 ‘H160 VIP - STC MAJOR Change -4+2 Conf. Introduc...</td>
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<td>21/06/2024</td>
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<tr>
<td>10080837</td>
<td>ALAMO ENGINEERING GmbH</td>
<td></td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Bombardier CL-600-2B16 as PED Tolerance This design change covers the approval for aircraft PED tolerance to use Transiently Portable Electronic Devices (T-PEDs) during all phases of flight except low visibility approach operation (ILS CAT II) on all models CL-600-2B16 (604 variant). The installation...</td>
<td>The Certification Basis for the original product as amended by other previously installed or repair.</td>
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<td></td>
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<td>07/12/2022</td>
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<td>10080837</td>
</tr>
<tr>
<td>10080840</td>
<td>AEROTEC &amp; CONCEPT</td>
<td></td>
<td>F900 EVASAN Stretchers Installation/Installation of four (4) Universal Convertible Stretchers, four (4) Oxygen Racks (without oxygen bottles) and relative cabin adaptation.</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by other previously installed or repair.</td>
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<td></td>
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<td>07/12/2022</td>
<td>Active</td>
<td>10080840</td>
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<td>STC Holder</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>777F, 777-300ER</td>
<td>Rotates film on fuselage/application of self-adhesive film on the fuselage. A microstructure pattern is applied to the film in order to influence the boundary layer and reduce drag. Flap 1: Removal of certain limitations (FWS CG) and update of ICA.</td>
<td>The Type and OSD Certification Bases (CB) for the original production chemises applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Supplemental Type Definition Document STCD11-1/201M Issue 03 rev 00 - Airplane Flight Manual Supplement - A-00-2202-APN Issue 08 rev 00 - Instructions for Continued Airworthiness A-00-2226 - ICA Issue 13 rev 00. Later revisions of the above listed document(s) approved/accredited the EASA system.</td>
<td>09/04/2024</td>
<td>Active</td>
<td>1200640</td>
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<tr>
<td>ELBE FLUGZEUGWERKE E Gmbh</td>
<td>A360 B4-622, A300 B4-622 R</td>
<td>Exchange of Emergency Locator Transmitter (ELT) on A300B4-600 F17. This modification allows the exchange of the existing fixed/portable ELT by another fixed/portable ELT. The proposed installation substantiates that the requested ELT and the containing non-rechargeable Lithiumbatteries ensures function.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): CRU_F-0301-02 _Non-rechargeable Lithium batteries installations - SC The requirements for environmental protection and the associated certification basis.</td>
<td>- Certification Plan ref. EFPI-TP-0146-001 Rev. C - Engineering Order ref. EFPI-ED-0146-001 Rev. A - Justification Report for ELT Installation Ref. EFPI-E- ED-0087 Rev. 0. Latest revisions of the above listed document(s) approved/accredited the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/12/2022</td>
<td>Active</td>
<td>1200650</td>
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<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>1900D</td>
<td>Installation of TCAS II system and replacement of mode-S transponder</td>
<td>The Type and OSD Certification Bases (CB) for the original production chemises applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>- Installation Instructions, document AT-006-INS, Rev. 0, dated 19.11.2022 - Aircraft Flight Manual Supplement, document AT-006-AMF, Rev. 0, dated 19.11.2022 - Instructions for Continued Airworthiness, document AT-006-ICA, Rev. 0, dated 19.11.2022 later revisions of the above listed document(s) approved/accredited.</td>
<td>The installed TCAS II system (Version 7.0) must be obligatory upon the airworthiness of the product.</td>
<td>12/12/2022</td>
<td>Active</td>
<td>1200662</td>
<td></td>
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</tr>
<tr>
<td>AIR TEAM SERVICE S.R.O.</td>
<td>208B 208</td>
<td>Installation of research equipment for atmospheric analysis theercraft cabin. The equipment consists of a VOC analyzer tIgonTR-PRTR-TOF 6000Q2 with CHARON extension module, vacuum generation unitQ15 with sound enclosure kitXK15, DC 28V to AC 230V converter, and mounting racks.</td>
<td>The Type and OSD Certification Bases (CB) for the original production chemises applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installation Instructions, document ATS-006-INS, Rev. 0, dated 19.11.2022 - Aircraft Flight Manual Supplement, document AT-006-AMF, Rev. 0, dated 19.11.2022 - Instructions for Continued Airworthiness, document AT-006-ICA, Rev. 0, dated 19.11.2022 later revisions of the above listed document(s) approved/accredited.</td>
<td>Applicable only to Cassapa 208B S/N 208B2430 with installed STC EASA.A.S.03334. Installation of Camera Kit. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect.</td>
<td>12/12/2022</td>
<td>Active</td>
<td>1200663</td>
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<tr>
<td>GDC ENGINEERING GmbH</td>
<td>A321-251N</td>
<td>Aircraft model applicability corrected.</td>
<td>The Type and OSD Certification Bases (CB) for the original production chemises applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Certification Document List ref. CCA- A321A-MCDL Issue 01, issued December 07. 2022 - Aircraft Flight Manual Supplement, document AT-006-AMF, Rev. 0, dated 01.06.2022.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/12/2022</td>
<td>Active</td>
<td>1200664</td>
<td></td>
<td></td>
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</tbody>
</table>
10080865
GIVH AEROSPACE (IRELAND) LIMITED
A340-643 A340-642
A340-600 Main Cabin Cargo System
Conversion This design change is to permanently convert the current passenger cabin to a Class F cargo compartment. The pre-mod configuration has a standard passenger cabin above the floor, and conventional cargo compartments below the floor. The post-mod configuration will hav...

10080867
BOGGI AERONAUTICS S.R.L.
429
Mission UHF, HF Radios and antennas Installation
The Type and OSD Certification Basis (CB) for the original products/applications applicable to this certificate/approval.The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...

10080895
PMV ENGINEERING
A340-542 A340-541
A340-500 cabin reconfiguration for GLOBAL
The Type and OSD Certification Basis (CB) for the original products/applications applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...

10080900
THALES AVIONICS Inc.
FAA STC ST02798LA
777F 777-300, 777-300ER 777-200, 777-200LR
Installation of the upgraded Thales Radio Attitude Transceiver Unit,ERTS
The Certification Basis for the original product as amended by the following or alternative airworthiness requirements:

10080902
ALAMO ENGINEERING GmbH
DORNIER 328-100
Leading Gear Warning Inhibit Certification of the installation of a guarded audio inhibit switch on the outboard instrument panel that allows the pilot to mute the low-level warning sirens provided by the landing gear system and audio callouts of the previously installed EGPWS system. The original TAWS...

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

The following paragraph(s) at a later amendment:

The following related design changes are prerequisite for this installation:
<table>
<thead>
<tr>
<th>STC Holder</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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</thead>
<tbody>
<tr>
<td>AEROLITE AG</td>
<td>MBB-BK117 D-3 MBB-BK117 D-3 MBB-BK117 D-2</td>
<td>Introduction of the Emergency Medical Service (EMS) equipment Part Number (PN) 145-0320030-501 (also referred to as &quot;CabinLayoutConfiguration A&quot;) installed in the helicopter cabin (Aerolite reference#131655) Revision 1</td>
<td>The Type and OSD Certification Bases (CB) for the original products/achievements applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Program CPR-22-038 Rev. B dated 15/12/2022 (v) Flight Manual Supplement FMS-145-0320030-501 Rev. 1 dated 22/11/2022 (v) Operations Manual, including Instruction for Continued Maintenance Airworthiness OpM-145/03 32030-501 Rev. 1 dated 22/11/2022 (v) Master Data List MDL-22-004 Rev. 1C.</td>
<td>The following are pre-requisite installations for the embodiment of this design change: - Airbus Helicopters Deutschland (AHD) floor (*) and FWD RH and LHaft-facing AHD seats (**) for the floor protection installation P/N 145-004723-505 - Floor protection installation P/N 145-004723-505 for the floor...</td>
<td>17/11/2023</td>
<td>Active</td>
<td>10080916</td>
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<tr>
<td>DOA AESGLOBAL LIMITED</td>
<td>737-800</td>
<td>INTRODUCTION OF REPLACEMENT G4 GAILEY'S CHANGE INTRODUCES A REPLACEMENT GALLERY IN THE G4 POSITION ON THE AIRCRAFT</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25.460.25.816.25.812.25.1309.25.1310.25.1311.25.1332.25.1333.25.1334.25.1341-CS 25.12.27 : CS 25.12.27</td>
<td>Master Data List: AE 707-7057 MDL Issue 01 CAA Supplement: AES-3456 Issue 015MM, IPC and WDM Supplements: AES-TP-042 Issue 01/50 later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>The aircraft must be maintained in accordance with ICA supplement/reference AES- TR-3456 Issue 01. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>16/12/2022</td>
<td>Active</td>
<td>10080919</td>
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<tr>
<td>THE NORDAM GROUP LLC</td>
<td>FAA STC ST09928AC</td>
<td>EFSYS Nose Radome Assembly Installation of an Enhanced FlightVision System (EFSYS) Nose Radome assembly installation of a Rockwell Collins (Collins Aerospace) EFS-4800 sensor (infrared camera) Revision 1: Added reference to FAA STC FMS 10-DRF-062 issue 1 Flight Manual supplement CA-117-25-50-00084 Issue 1 Instructions for continued compliance Ref-MD-117-25-50-00083 - Master Data List MDL-01 rev. B dated 08 April 2022</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25.460.25.816.25.812.25.1309.25.1310.25.1311.25.1332.25.1333.25.1334.25.1341-CS 25.12.27</td>
<td>Master Data List ND000- MDL01 rev. B dated 08 April 2022</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>13/01/2023</td>
<td>Active</td>
<td>10080937</td>
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<tr>
<td>DRF STIFTUNG LUFTRETTUNG EMMENJUETZIGE AG</td>
<td>MBB-BK117 C-2, MBB- BK117 D-2</td>
<td>Powered Stretcher System</td>
<td>The Type and OSD Certification Bases (CB) for the original products/achievements applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master document List ref- MD-117-25-50-0083 MDL01 rev. B dated 08 April 2022</td>
<td>Master Data List: N “” 1568 Issue 3(Reconfiguration of Compliance Ref: DC-117-25-50-0003, Issue 0) Installation Manual - IM-117-25-50-0003 Issue 1 Basic and EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>21/12/2022</td>
<td>Active</td>
<td>10080942</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>AS 332 L, AS 332 L1 AS 332 C, AS 332 C1</td>
<td>Lifesaver Installation Rev. 1 corrects the Technical Documentation List.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: FAR 29 20.20 for 06/1926. The requirements for environmental protection and the associated certified noise regulations...</td>
<td>Certification Program 21430/ACP, Revision 04, dated 24/11/2022 System Technical Specification 2143-01-A-STS, Revision 01, 23/09/2022 Master Data List 2143-01-A-MDL, Revision 01, dated 16/12/2022.</td>
<td>- STC is limited to the installation variant 2 as defined in document 2143-01-A-STST. Revision 01 - The modification MOD 2143-02 is a pre-requisite for the embodiment of the design change. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previous...</td>
<td>21/12/2022</td>
<td>Active</td>
<td>10080567</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>10080659</td>
<td>SAFRAN ENGINEERING SERVICES GmbH</td>
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<td>EC 225 LP</td>
<td>Night Vision Imaging System (NVIS) capability on EC 225 LP (Ref.MOD-H2001)</td>
<td>The Type and OSD Certification Bases for the original product as amended the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS-29 Initial Issue: The requirements for environmental protection and the associated certified noise and/or emissions levels of the product.</td>
<td>Certification Program MOD-H2001-CP, Issue 4, dated 22/08/2022/System Descriptive Document MOD-H2001-SDD, Issue 4, dated 20/09/2022/Rotorcraft Flight Manual Supplement MOD-H2001-RFM; Revision 1, dated 19/12/2022/</td>
<td>20/12/2022</td>
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<tr>
<td>10080972</td>
<td>AEROMECH Inc.</td>
<td>FAA</td>
<td>ST037235E</td>
<td>737-800</td>
<td>RVSM Group Approval for Boeing 737-800 BDSF Aircraft</td>
<td>The Type and OSD Certification Bases for the original product as amended the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: EASA CS/ACNS Subpart E, Section 2 (RVSM) Issue 3. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product.</td>
<td>MASTER DATA LIST AM0375-DL02-00; 21 October 2022/FM SUPPLEMENT AM0375-FM01-00r later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>21/12/2022</td>
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<tr>
<td>10080977</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td></td>
<td></td>
<td>A320-171X/172X/171NX/211X/212X/251X/253N/251NX/252X/253</td>
<td>Activation of an Air Cleaning System (ionization) This change includes the activation of an installed or installed system work on the ionization principle.</td>
<td>The Type and OSD Certification Bases for the original product as amended the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: The requirements for environmental protection and the associated certified noise and/or emissions levels of the product.</td>
<td>The change is defined with Revision A of Master Data List document/MIL-ALLAC01-210. The aircraft shall be maintained in accordance with Revision A and its instructions for Continued Airworthiness document ICA-ALLAC01-210r later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that there will be no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10080861</td>
<td>AMTECH AERONAUTICAL LIMITED</td>
<td></td>
<td></td>
<td>TCCA</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Pratt and Whitney Engine Services, Inc. Flight Data Acquisition Storage and Transmission System (FAST).</td>
<td>Master Data List MDL463631, Revision 3, 30 October 2018/Supplement AFMS GN4-T-250-7 Revision B, 18-05-22r later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU Master Data Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that there will be no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10080864</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA</td>
<td>ST043106Y</td>
<td>737-800</td>
<td>Installation of a Cabin Interior</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS-25 and/ or 25.820a, 25.835a, 25.86a, 25.88a. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product.</td>
<td>Delta Engineering Master Data List 0106-10378-2429, Revision T, dated 8/November 2022/</td>
<td>06/02/2023</td>
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<td>EASA STC no.</td>
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<td>ST03611NY</td>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>ATR 72-212 A</td>
<td>Installation of a Service Deck</td>
<td>04/01/2023</td>
<td>Active</td>
<td>10080985</td>
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<td>ATR Cargo Compartment Doublets</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: JAR 25.1, 25.20 (b), 25.301 (a) – (b), 25.303, 25.305 (a) – (b), 25.307 (a), 25.561 (b), 25.571 (a), 25.601...</td>
<td>Not applicable to ATR 72-212A with MOD 7900 installed. Prior to installation of this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/01/2023</td>
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<td>10080985</td>
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<tr>
<td>ST03611NY</td>
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<td>ATR 72-201, ATR 72-202</td>
<td>Replacement of Rockwell Collins LRA-990 Low Range Radio Altimeters with upgraded LRA-990 units for improved RF immunity to 5G cellular data interference</td>
<td>FAA Approved Model List (AML) RAC-0696, Revision (-), dated February 26, 2019, Instructions for Continued Airworthiness of Aircraft (ICA) DM1518-5, Revision (A), dated February 26, 2019, or later revisions of the above listed document(s) approved/accepted by EASA.</td>
<td>11/01/2023</td>
<td>Active</td>
<td>10080990</td>
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<tr>
<td>ST03811NY</td>
<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FALCON 7X</td>
<td>Installation of Garmin GI 275 Standby Flight Display System (SFDS) nDassault Aviation model Falcon 7X aircraft</td>
<td>Applicable to Dassault Aviation Falcon 7X aircraft serial numbers 001 through 400 and Falcon 7X aircraft on which modification 01000 and 01254 are installed at production (Falcon 8X Definition), aircraft numbers 0401 and subsequent. Software installed in GI-275 units must conform to Hardware &amp; S...</td>
<td>09/01/2023</td>
<td>Active</td>
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<td>AVIATION ENGINEERING AS</td>
<td>ST01972WI</td>
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<td>DUNCAN AVIATION, Inc.</td>
<td>ST019072WI</td>
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<td>FALCON 2000, FALCON 2000EX</td>
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<td>MED-PAC INC.</td>
<td>ST034555CH</td>
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<td>FALCON 2000, FALCON 2000EX</td>
<td>Installation of Installation of Clean Air Ionization System &amp;</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>A340-313</td>
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<td>A340</td>
<td>Installation of Installation of Clean Air Ionization System &amp;</td>
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<td>MT-PROPELLER ENTWICKLUNG GMBH</td>
<td>T67M260 FIRELY</td>
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<td>T67M260 FIRELY</td>
<td>Installation of Installation of Clean Air Ionization System &amp;</td>
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<td>10081042</td>
<td>SAFRAN CABIN INC.</td>
<td></td>
<td>ST00484SE</td>
<td>A320-200 CEO</td>
<td>Installation of interior retrofit kit in the passenger cabin.</td>
<td>Health Tecna inc. Drawing List HPV-DL-55, Revision AU, dated 26 October 2002 &lt;/p&gt;</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/04/2023</td>
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<td>10081043</td>
<td>SAFRAN CABIN INC.</td>
<td></td>
<td>ST00773SE</td>
<td>A320-200 CEO</td>
<td>Installation of Upper Deck Overhead Stowage Bin Interior Retrofit.</td>
<td>Health Tecna, Inc. Drawing List HPV-DL-71, Revision T, dated 9 March 2004 &lt;/p&gt;</td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/05/2023</td>
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<td>10081046</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 1</td>
<td></td>
<td>777-300ER</td>
<td>4UJ Cabin Reconfiguration.IFE Zodiac; “RAVE Centric” System InstallationREV. 1 adds reference to Change Approval Sheet which includes list of definition and substantiation data.This revision also adds reference to accomplishment Service Bulletin for modification.</td>
<td>Change Approval Sheet CAS-777-19-003 Rev. 00, Service Bulletin 4FB-777-19-003-05-001 Rev. 00, Certification Program CP-777-19-003 Rev. 5MEL Supplement MMELE-777-19-003-01 Rev. 00, Instructions of Continued Airworthiness ICA-777-19-003 rev 0 or later revisions of the above listed document(s) approved/accredited indefeasibly of EASA in accordance with the Technical Implementation Procedures.</td>
<td>- Aeronaves modifiadas per esta STC are considered to be a “variant” to the 777-300ER baseline configuration. Cabin crewoperating aeroplanes modified per this STC must complete aircraft difference training in accordance with ORD CC.130, which must include the approved OSD-Cabin Crew Training Document (CCD) - Boeing...</td>
<td>18/01/2023</td>
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<tr>
<td>10081048</td>
<td>BOEING IRELAND LIMITED</td>
<td></td>
<td></td>
<td>787-8</td>
<td>355 Passenger Interior Reconfgurati...</td>
<td>The change is defined with Revision 00 of Boeing Ireland Change Approval document BL-00235-APP.The aircraft shall be maintained in accordance with Revision 00 of Boeing Ireland Limited Instructions for Continued Airworthiness document BL-00235-ICA after later revisions of the above listed document(s) approved...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/01/2023</td>
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<td>10081050</td>
<td>GDC ENGINEERING GmbH</td>
<td></td>
<td>A320-214</td>
<td>A320-200 CEO</td>
<td>The Type and OSD Certification Bases (CB) for the original product are applicable to this certificate/approval. The requirements for environmental protection and the associated noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval w...</td>
<td>Description Report &amp; LOI (Certification Plan) A320G-190-25000 Issue Dated 08 December 2002 Master Certification Document List CCN-3320-MCDL Issue (1 dated 13 January 2023) A3111-ICA-KME Rev 00 dated 12 December 2022/4 later revisions of the above listed document(s) app...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>17/01/2023</td>
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10081075 HONEYWELL INTERNATIONAL Inc. REV. 2 FAA ST02796LA 767-300F, 767-400ER 767-200, 767-300 Upgrade of Radio Attenuator System to install Honeywell ALA-52B SystemRev. 1- Editorial changeRev. 2 corrects a typo in the Description of Design Change.

The Certification Basis for the original product as amended by the following paragraph(s): at a later amendment: C525 and/or C525B. The requirements for environmental protection and the associated certified noise level.

Certify Nation Master Data List, document number C102-024MDL-00.Revision 0, dated 16 October 2022; (i.e.) Instructions for Continued Airworthiness, Certify Nation document number NC102-024IC-01. Revision 0, dated 14 October 2022; (i.e.) for later revisions of the above listed document(s) approved/accepted:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effect upon the airworthiness of the product.

09/02/2023 Active 10081075

10081077 DOA AESGLOBAL LIMITED REV. 1 737-800 INTRODUCTION OF REPLACEMENT GT GALLEY THIS CHANGE INTRODUCES A REPLACEMENT GALLEY IN THE G7 POSITION ON THE AIRCRAFT

The Type and OSO Certification Bases for the original product as amended by the following paragraph(s): at a later amendment: C525 and/or C525B. The requirements for environmental protection and the associated certified noise level.

Master Data List AES-737-6597MDL Issue 01CA supplement: AES-TR-3410 Issue 01AMM, IPC and WDM Supplements: AES-TR-041Issue 01for later revisions of the above listed document(s) approved/accepted/under the EASA system.

The aircraft must be maintained in accordance with ICA supplement reference AES-TR-3410. This STC must be performed concurrently with approved Minor Modifications:

AES-737-6592; AES-737-6632; AES-737-6693; AES-737-6696; AES-737-699P. Minor Modifications required to installation of this change/repair must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effect upon the airworthiness of the product.

20/01/2023 Active 10081077

10081079 AIRBUS S.A.S. A330-743L MOD A60210 - MODIFY EMERGENCY EQUIPMENT ARRANGEMENT FOR OVERWATER OPERATIONS

The Type and OSO Certification Bases for the original product as amended by the following paragraph(s): at a later amendment: C525 and/or C525B. The requirements for environmental protection and the associated certified noise level.

Modification Approval Sheets ref: MAS A60210 issue 01-Certification Summary Report ref: SRSA-22-006 Issue 04 for later revisions of the above listed document(s) approved/accepted/under the EASA system.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will not introduce any adverse effect upon the airworthiness of the product.

18/01/2023 Active 10081079


The Type and OSO Certification Bases (CB) for the original product as amended by the following paragraph(s): at a later amendment: C525 and/or C525B. The requirements for environmental protection and the associated certified noise level.

To be operated in accordance with MT-Propeller AFM-S Document No. E-3989Rev. 0To be installed i.a.w. MT-Propeller Installation Instruction Doc. No.E-3989 Rev. 0To be maintained i.a.w. MT-Propeller ICA Doc. No. E-3990 Rev. 0To be maintained.

According to limitation section of AFM(S) prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce any adverse effect upon the airworthiness of the product.

19/01/2023 Active 10081080

10081086 ADAC HELISERVICE GMBH MBB-BK117 D-3 MBB-BK117 D-2 Installation of retractable LED search & landing light Luminitor LS400M(LH and RH) in replacement of pre-installed Zodiac Aero Electronictactable search & landing light, as well as associated Night Vision Imaging System (NVIS) approval

The Type and OSO Certification Bases (CB) for the original product as amended by the following paragraph(s): at a later amendment: C525 and/or C525B. The requirements for environmental protection and the associated certified noise level.


Limited to helicopters with Helionix cockpit variant. As prerequisite (pre-modification configuration) the Rotorcraft must be equipped with the following retractable search and landing (LS) lights: Zodiac Aero Electric S/L, light P/N 4361849/ AND P/N D3334242101(i.a.w. Airbus IPC 33-43-03-01A – RH...
List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<th>Rev. level</th>
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<tr>
<td>10081088</td>
<td>HAECO CABIN SOLUTIONS, LLC</td>
<td>REV 1</td>
<td>FAA ST0457AT</td>
<td>737-800</td>
<td>Interior Recomfiguration installation in a Boeing 737-800 Seriesaircraftliner. 1 adds the FAA STC number.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.1031(a)(36): 27: 25.853 (a)(8)(iii)(b): 20.981 (d)(7)(i)The requirements for environmental protection and the associated certificated...</td>
<td>HAECO Cabin Solutions Master Data List 2170606L01, Revision R, dated 28 Dec 2022. 1(S) &gt;HAECO Cabin Solutions Instructions for Continued Airworthiness 2170606R09, Revision 00, dated 11 Aug 2022. 1(S) &gt;or later revisions of the above listed document(s) approved/ accepted on behalf of EASA in accordance with...</td>
<td>Prior to installation of this change/repair it must be determined that the interf...</td>
<td>06/02/2023</td>
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<td>10081097</td>
<td>STANDARD AERO AVIATIONSERVICES ES, L.L.C.</td>
<td></td>
<td>FAA STC ST04547CH</td>
<td>GIV-X</td>
<td>Installation of a Viasat Ka SATCOM</td>
<td>The Certification Basis for the original product remains applicable if the following instructions are not applied:</td>
<td>1. Master Data List 1040246. Revision (D), dated 16 Nov 2022. 2, Airplane Flight Manual Supplement, 1040197, Revision (I), dated 07 Nov 2022. 2. Instructions for Continued Airworthiness, Report No. 1040287. Revision (A), dated 26 Oct 2022. 2(S) &gt;or later revisions of the above listed document(s) approved/ accepted on behalf of EASA in accordance with...</td>
<td>This STC is applicable to all serial numbers for these Make-Models. EASA...</td>
<td>23/01/2023</td>
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<td>10081099</td>
<td>STANDARD AERO AVIATIONSERVICES ES, L.L.C.</td>
<td></td>
<td>FAA STC ST04544CH</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of a Viasat Business Aviation Router (VBR) P/N 137138</td>
<td>The Certification Basis for the original product remains applicable if the following instructions are not applied:</td>
<td>1. Master Data List 1040239. Revision (E), dated 15 Nov 2022. 2, Airplane Flight Manual Supplement, for each Make-Model as listed on the FAA AML number ST04544CH. 3. Instructions for Continued Airworthiness, for each Make-Model, as listed on the FAA AML number ST04544CH or later revisions of the above listed document(s) approved/ accepted on behalf of EASA in accordance with...</td>
<td>This STC is valid for aircraft types and models approved by EASA and listed on the FAA STC ST04543CH dated 16 November 2022. This STC is applicable to all serial numbers for these Make-Models. This STC requires prior and concurrent embodiment of a TPED STC service Bulletin to ensure...</td>
<td>23/01/2023</td>
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<td>10081114</td>
<td>AVIATION DEVICES &amp; ELECTRONIC CO. I.PONENTS LLC (Av-DEC)</td>
<td></td>
<td>FAA STC ST02532LA</td>
<td>SEE FAA AML</td>
<td>Modification of the wing fuel tank access covers by replacing the existing rubber strip and seal with a new encapsulated wire mesh gasket</td>
<td>The Certification Basis for the original product remains applicable if the following instructions are not applied:</td>
<td>Master Data List, K3176- MDL, Revision C, dated 17 November 2017. 1(S) &gt;Installation Instructions, Document Number 3L, FG_010: revision E, dated 7 August 2017. 2(S) &gt;AWL document as specified in AML-STC No. ST02532LA. 3(S) &gt;or later revisions of the above listed document(s) approved/ accepted on behalf of EASA in accordance with...</td>
<td>Prior to installation of this change/repair it must be determined that the interf...</td>
<td>24/01/2023</td>
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<td>10081120</td>
<td>AVIATION BUSINESS CONSULTANTS INTERNATIONAL S.R.L.</td>
<td></td>
<td>A320-253N/-271N/-272N/-273N</td>
<td>LOPA modification on A320 ACThis design change is aimed to perform passenger layout re-configuration of the A320 AC type by replacing the existing passenger seats with the ‘Essenza’ D3 seats model.</td>
<td>The Type and OSD Certification Bases (CSIs) for the original product remain applicable to this certificate/ approval: The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval...</td>
<td>- SB-CNG00022- DF-003-00-E, Rev. 0(S) &gt; Service Bulletin Installation- 529-0022-1000, Rev. 0(S) &gt; CABIN SEAT LAYOUT/TLOPA modification- MMS- CNG00022-DF-030-30-G, Rev. 0(S) &gt; Maintenance Manual Supplement- AML- CNG00022-DF-030-01-E, Rev. 0(S) &gt; Approved Drawing List later revisions of the above...</td>
<td>Prior to installation of this change/repair it must be determined that the interf...</td>
<td>27/01/2023</td>
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<td>10081140</td>
<td>AVIATION SPECIALTIES UNLIMITED, INC</td>
<td>FAA SR016115E</td>
<td>MBB-BK117 D-2, MBB-BK117 D-3</td>
<td>MBB-BK117 C-1, MBB-BK117 C-2, MBB-BK117 B-1, MBB-BK117 B-2, MBB-BK117 A-3, MBB-BK117 A-4</td>
<td>EASA validation of the Night Vision Imaging Systems (NVIS) FAA/Supplemental Type Certificate (STC) ref. SR016115E</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Project Specification Certification Plan (PSCP) ref. ASU-PSCP-BK117D-2112/Revision C, dated 23/09/2022; Rotorcraft Flight Manual (RFM) Supplement ref. FMS-117004-A Revision R, dated 26/10/2022 and approved by the FAA on 02/11/2022; Instructions for Continued Airworthiness (ICA) document ref. ICA-117004-A Rev. 2. This design change is limited to MBB-BK117A-3/A-4/B-1/2/C-1/C-2/G-3/G-9 helicopters with a configuration described in the ICA-117004-A Appendices Revision IR, dated 15/09/2022, supplementing the ICA document ref. ICA-117004-A Revision H, dated 06/05/2022. Rotorcraft modified by this design change.</td>
<td>26/01/2023</td>
<td>Active</td>
<td>10081140</td>
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<tr>
<td>10081145</td>
<td>Deutsche Aircraft GmbH</td>
<td>DORNER 328-100</td>
<td>IFOD – In-Flight Operable Door</td>
<td>This change replaces the original Cargo Door by an In-Flight Operable Door.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): ORI D-1-11 IFOD – In-Flight Operable Door Issue 3. The following paragraph(s) at a later amendment: CS 25.783 Amitd. 27. The requirements...</td>
<td>Master Document List, Doc.-No. MDL 1792/00, Issue A, Date 26.01.2023; Aircraft Flight Manual Supplement, Doc.-No. AFMS12B1792/00, Issue A, Date 19.01.2023; Airplane Maintenance Manual Supplement, Doc.-No. AAM512B1752/0001, Issue A, Date 24.11.2022 (containing an Airworthiness Limitation Section).</td>
<td>This scope of this STC is limited to the installation of the door in the aircraft and the operation of the door in flight (open—close). Additional operations like the dropping of stores from the aircraft are out of the scope of this STC. Therefore, a separate operational approval may be required from 1..</td>
<td>27/01/2023</td>
<td>Active</td>
<td>10081145</td>
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<tr>
<td>10081148</td>
<td>BOEING IRELAND LIMITED</td>
<td>787-9</td>
<td>Boeing 787-9 – Premium Economy Introduction</td>
<td>The aircraft are configured in Zones B and C in order to introduce an additional Economy Class. The configuration of the aircraft is unchanged.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Change / Repair Approval, Boeing 787-9 – Premium Economy Introduction, BIL-00185-APP. Revision W, dated 24th January 2023; (containing an Airworthiness Limitation Section).</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>27/01/2023</td>
<td>Active</td>
<td>10081148</td>
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<tr>
<td>10081151</td>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>INSTALLATION OF PALLETS COUPLERS TO CONNECT TWO (ETIS0-C050) pallets, NAA/S3610/A3610100 size code A or M, installed longitudinally in a direction double the length.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.783 Amitd. 27. The requirements...</td>
<td>Certification Documents, Ref. 3000096 DOC, Is. 1, Date 25.01.2023; Installation Documents, Ref. 3000095 STC, Is. 1, Date 25.01.2023; Engineering Order, Ref. 3000095-EO, Is. 1, Date 23.01.2023; Component Maintenance Manual, Ref. 3000095-CMM, Rev. 1, Date 30.11.2022.</td>
<td>Installation shall be done within the limitations identified for the specific aircraft type in the related Weight &amp; Balance and Operation Manual(s). See Component Maintenance Manual, Ref. 3000095-CMM for installation of this change/repair it must be determined that the interrelationship between this...</td>
<td>30/01/2023</td>
<td>Active</td>
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<tr>
<td>10081171</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING SERVICES, INC, WALTER STARZACHER</td>
<td>A320-272N</td>
<td>180 Passenger Seating Layout</td>
<td>This modification introduces a cabin layout using 60 economy class seats generating a 180 passenger seating configuration.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.1, CS 25.29, CS25.301 (a-b), CS 25.303+ (n-v)/CS 25.305 (a-b), CS 25.307 (a), CS 25.321, CS 25.337...</td>
<td>The change is defined with revision 1 of AMES Engineering Order Document EO M398010016, JTU _23092022Maintenance shall be accomplished in accordance with revision 1 of AMES/Order for Continuous Airworthiness Document/ICA_M38900106_0_JTU _23092022 latter revisions of the above listed document(s) a...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/02/2023</td>
<td>Active</td>
<td>10081171</td>
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<td>STC Holder</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<tr>
<td>KAMATICS CORPORATION</td>
<td>Installation of Garmin GA 36 GPS/WAAS antenna and Bendix/King AeroWave/GA antenna.</td>
<td>- MDL Document Number KAM-3587-MDL-001 at Revision 8, dated July 27, 2022, Installation instructions SI Document NumberSIN3587 at Revision 8, dated July 11, 2022.</td>
<td>- ICA Document Number KM-3587-ICA-001 at Revision B, dated July 1, 2012. - Airworthiness Limitations Section (ALS) of ICA Document NumberICA..</td>
<td>Only structural aspect of the installation of antennas is approved. Thesystemic interfaced with aircraft systems is not approved by the STC. Prior to installation of this change/repair, it must be determined ifthe interface between this change/repair and any other previously installed equipment is appropriately addressed.</td>
<td>02/02/2023</td>
<td>Active</td>
<td>10261194</td>
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<tr>
<td>AVONIK STRAUßING ENTWICKLUNGS GmbH</td>
<td>Installation of new Ignition system for TB10 O-360 engine. Replacement of theeexisting D4LN0000 dual mageto by the combination of four single Sics 4373 magneto and an electronic ignition system. manufactured by Elektoral Rev. 1: Proposing a new ENAC designed Trigger Wheel and a more secure ENAC designed command panel.</td>
<td>- Compliance Program EAD4_MDC_EASA, Rev. R0 dated 31st January 2023, Declaration of compliance EAD4_DUC_EASA, Rev. 0 dated 31st January 2023.</td>
<td>- Installation instructions EAD4_INST_AV169_EASA, R0 dated 29th April 2021. Flight Manual supplement EAD4_RFSMS_EASA, Rev. 0 dated 23 August 2022.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 21A, 109. Prior to installation of this change/repair, it must be determined if the interface between this change/repair and any other previously installed equipment is appropriately addressed. The introduction of a new hardware/software component between this change/repair and any other previously installed equipment may introduce an adverse effect upon the system.</td>
<td>06/08/2023</td>
<td>Active</td>
<td>10261196</td>
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<td>ENAC</td>
<td>New Ignition systems for TB10 O-360 engine. Replacement of the existing D4LN0000 dual magneto by the combination of four single Sikas 4373 magneto and an electronic ignition system. manufactured by Elektoral Rev. 1: Proposing a new ENAC designed Trigger Wheel and a more secure ENAC designed command panel.</td>
<td>- Compliance Program EAD4_MDC_EASA, Rev. R0 dated 31st January 2023, Declaration of compliance EAD4_DUC_EASA, Rev. 0 dated 31st January 2023.</td>
<td>- Installation instructions EAD4_INST_AV169_EASA, R0 dated 29th April 2021. Flight Manual supplement EAD4_RFSMS_EASA, Rev. 0 dated 23 August 2022.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 21A, 109. Prior to installation of this change/repair, it must be determined if the interface between this change/repair and any other previously installed equipment is appropriately addressed. The introduction of a new hardware/software component between this change/repair and any other previously installed equipment may introduce an adverse effect upon the system.</td>
<td>06/08/2023</td>
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<td>10261196</td>
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<td>ECSM AVIATION SYSTEMS GMBH</td>
<td>Installation of external load system EAD04.</td>
<td>- Compliance Program EAD4_MDC_EASA, Rev. R0 dated 31st January 2023, Declaration of compliance EAD4_DUC_EASA, Rev. 0 dated 31st January 2023.</td>
<td>- Installation instructions EAD4_INST_AV169_EASA, R0 dated 29th April 2021. Flight Manual supplement EAD4_RFSMS_EASA, Rev. 0 dated 23 August 2022.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 21A, 109. Prior to installation of this change/repair, it must be determined if the interface between this change/repair and any other previously installed equipment is appropriately addressed. The introduction of a new hardware/software component between this change/repair and any other previously installed equipment may introduce an adverse effect upon the system.</td>
<td>07/02/2023</td>
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<td>10261191</td>
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<td>NORTHWEST AEROSPACE TECHNOLOGIES NC.</td>
<td>Cabin Reconfiguration 330 passengers for Austrian Airlines Rev. 1: Approval of amended AFM Supplement for ELMS Load Shidding related procedure.</td>
<td>- Certification Summary N2070-D30 Rev. 0 Master Drawing List N2070-D10 Rev FICa Principle Manual N2070-D70 Rev AFM Supplement N2070-D90 Rev AF or later revisions of the above listed document(s) approved/ accepted/ adhered to the EASA system.</td>
<td>- The installations of the configuration covered by this STC have the following prerequisite installations - STC 10043471 (FAA/No. ST1025755) or - STC 10063845 + STC 10063846 + STC 10063846 + STC 10078270 or STC 10063845 + STC 10078270 Prior to installation of this change/repair it must be determined if the interface between this change/repair and any other previously installed equipment is appropriately addressed.</td>
<td>The approval holder shall fulfill the obligations of Part 21, Point 21A, 109. Prior to installation of this change/repair, it must be determined if the interface between this change/repair and any other previously installed equipment is appropriately addressed. The introduction of a new hardware/software component between this change/repair and any other previously installed equipment may introduce an adverse effect upon the system.</td>
<td>07/02/2024</td>
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<td>Models</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10081211</td>
<td>PEREGRINE LLC/O/a PEREGRINE</td>
<td></td>
<td>FAA STC SA01988WI</td>
<td>390</td>
<td>Installation to replace the existing De/Onoe single tail position anti-collision lights with Honeywell Aerospace Technologies Light Emitting Diodes (LED)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Operating Document No. EIV-21-0033, Revision B, signed 26 May 2022&gt;(v.3) &gt;or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of ELV USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/02/2023</td>
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<td>10081231</td>
<td>INFINION CERTIFICATION ENGINEERING</td>
<td></td>
<td>TCCA SA20-16, ISSUE 1</td>
<td>DHC-8-401, DHC-8-402 DHC-8-314, DHC-8-315 DHC-8-301, DHC-8-311 DHC-8-301, DHC-8-202 DHC-8-103, DHC-8-106 DHC-8-101, DHC-8-102</td>
<td>Flight Compartment Stowage Box: This stowage box is fabricated and installed in the aircraft, under the floor. This box is used for miscellaneous storage and is accessible from the flight compartment during flight.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment. For DHC-8-100, DHC-8-200 and DHC-8-300 models: FAR 25.1, 25.601,25.609, 25.1541 at Amdt 25-0, FAR 25.771 a...</td>
<td>Installation shall be performed in accordance with Infinion Certification Engineering Inc. Installation Instruction EIO-1051-2 Issue 0, Maintenance shall be performed in accordance with Infinion Certification Engineering Inc. Maintenance Manual Supplement and Instructions for Continued Airworthiness MMS...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/02/2023</td>
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<td>10081240</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td></td>
<td>FAA STC4344AT-D</td>
<td>GIV, GIV-X, GIV, GV, GV-SP</td>
<td>Installation of Satcom Direct Plane Simple KU System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Gulfstream Aerospace Corporation Index List GDL 10000064R, Revision (a), dated 01 December 2022&gt;(v.3) &gt;or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of ELV USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/02/2023</td>
<td>Active</td>
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<td>10081241</td>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td></td>
<td>FAA STC ST0133AT</td>
<td>767-300F 767-300 767-200</td>
<td>Installation of an Indium Communications System in B767-200/300/300F Series Aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Avionics Support Group, Inc. Master Data List 29126-MDL, Rev. 5, dated 28 September 2022&gt;(v.2) &gt;or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of ELV USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/05/2023</td>
<td>Active</td>
<td>10081241</td>
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<td>10081245</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td></td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>GV</td>
<td>Installation of L3 Harris GH-3900:2 Secondary Flight Display on Gulfstream Gv model aircraft</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>FAA Supplemental Type Certificate STC ST04338AT-D&gt;(v.3) &gt;Installation of L3 Harris GH-3900:2 Secondary Flight Display system 7 July 2021&gt;(v.3) &gt;or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of ELV USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/04/2023</td>
<td>Active</td>
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EASA STC No. | STC Holder | Rev. level | Foreign/NAA STC No. | Model | Description | Certification basis | Associated Technical Documentation | Limitations/Conditions | Date of Issue | Status | System Record
---|---|---|---|---|---|---|---|---|---|---|---
10081283 | DECA AVIATION ENGINEERING LIMITED | 03 | AS 355 N, AS 355 NP | AS 355.1, AS 355 F 1, AS 355 F 2 | Structural provisions and equipment installation of CCG-AGOS Survey System | The Type and OSD Certification Bases (CB) for the original product are applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w... | Modification Summary MS10096, Rev 10 | - For AS 355NP, CAT. 6 operations are not approved with the installation of this STC. This STC covers only the installation of equipment as items of mass. Separate approval is required for the operational, electrical, and functional aspects of the ACG Survey System installation and interconnection to... | 15/02/2023 | Active | 10081283
10081288 | HELI-ONE (NORWAY) AS | 03 | AS 332 L1 | AS 332 L | Required Navigation Performance (RNP) approach capability certification (Ref. modification AHSE-72850). | The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 29 Am 2 for the following requirement: 29.610. The requirements for environmental protection and the associated certified noise and/or emissions. | Certification Programme (CP) ref. AHSE-72852 Revision F, dated 01/11/2022; Flight Manual Supplement (FMS) for RNP approach capability ref. SUP H1 No.1. 340 Revision dated 23/01/2023. Product Change Declaration (PCD) ref. AHSE-72850 Revision F, dated 15/02/2023. Instructio... | This approval is only applicable to AS 332 L1 helicopter models equipped with AHAS (i.e. AS332 L1e). Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse... | 15/02/2023 | Active | 10081288
10081291 | ALAMO ENGINEERING GmbH | 03 | DORNIER 328-100 | Mission Cabin Configuration with 9 passenger seats | The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Am 27. The requirements for environmental protection and the associated certified noise and/or emissions. | Master Document List (MDL232A5E-01) with Rev. A dated 07/02/2023; Operator Manual (OM-A) with version 1.1 dated 15/08/2020. Later revisions of the above listed document(s) approved/accepted under the EASA system. | Limited to AIC with EASA STC 10013693 (former STC EASA A.O.1004) embodied. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness... | 16/02/2023 | Active | 10081291
10081295 | SABENA TECHNICS BGC | 03 | AS31-272N | A321 Cabin Reconfiguration from 150PA to 174PA (2141Y) | The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Am 27. The requirements for environmental protection and the associated certified noise and/or emissions. | Master Data List, Ref. 40916-01-A-MDL-R00 dated February 15th 2023; Instr... | As a concurrent requirement of this change, accomplishment of this STC requires the simultaneous accomplishment of the following modification: Installation of a photo-luminescent EFPM as per Luftfahrt Technik & Engineering Bulletin A320-ES33-0384, approved by EASA STC No. 10014013. Prior to installation... | 16/02/2023 | Active | 10081295
10081304 | BOOST HUMAN EXTERNAL CARGO SYSTEMS | 03 | TCCA SH15-39, ISSUE 8 | Dual Hook System (ref. B1310-ZHS) for Human External Cargo. | The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 27 Am 6. The requirements for environmental protection and the associated certified noise and/or emissions. | Master Document List, MDL-B1310-DHS, Rev. NC, dated January 26, 2023; Certification Plan CP1517, Rev. D, dated December 19, 2022; Flight Manual Supplement FMS-81310-EASA, Rev. NC, dated January 28, 2023; Installation Instructions, INSTALL- B1310 Rev. E, December 19, 2022; Instructions for Continued Airworth... | - Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness... | 21/02/2023 | Active | 10081304
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>JET AVIATION AG</td>
<td>GVI (G650)</td>
<td>Installation of new Cabin Layout 'WOLF' for Gulfstream GVI (G650) aircraft, in accordance with Jet Aviation Master Data List JBSL-M-03524ML-001. The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval. The following modification is a prerequisite for this installation: JBSL-M-03524ML-001, Issue 2, dated 17 Feb 2023.</td>
<td>Jet Aviation Master Data List JBSL-M-03524ML-001, Issue 2, dated 17 Feb 2023.</td>
<td>- The following modification is a prerequisite for this installation: JBSL-M-03524ML-001, Issue 2, dated 17 Feb 2023.</td>
<td>22/02/2023</td>
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<td>10081315</td>
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<td>NSE</td>
<td>MBB-BK117 D-3 MBB-BK117 C-2</td>
<td>Installation of the BCT communication and tracking unit EC145-H45 Ph N00057647AA. The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Certification Plan No. PCERT-0004966AA, Rev. ART/Roomcraft Fight Manual Supplement No. RFMS-BCT-EC145-EN, rev 01/Conformity Statement and Documentary architecture No. CONFAD-0004966AA, Rev A1, dated 10/04/2021.</td>
<td>- Certification Plan No. PCERT-0004966AA, Rev. ART/Roomcraft Fight Manual Supplement No. RFMS-BCT-EC145-EN, rev 01/Conformity Statement and Documentary architecture No. CONFAD-0004966AA, Rev A1, dated 10/04/2021.</td>
<td>24/02/2023</td>
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<td>TURK HAVA YOLLARI TEKNIK A.S. (TURKISH YOLLARI TEKNIK A.S.)</td>
<td>C0359 A321 High Density Cabin Modification of all 194/540 seats for Gulfstream G650 aircraft with 215 Safran Z85B-600037647AA.</td>
<td>C0359 A321 High Density Cabin Modification of all 194/540 seats for Gulfstream G650 aircraft with 215 Safran Z85B-600037647AA.</td>
<td>C0359-AWG01-00 LOPA Drawing C0359-DWG01-00. Emergency Equipment Layout/ Y35900359-DWG03-00 Emergency Equipment Layout/ Y35900359-DWG03-00.</td>
<td>C0359-AWG01-00 LOPA Drawing C0359-DWG01-00. Emergency Equipment Layout/ Y35900359-DWG03-00 Emergency Equipment Layout/ Y35900359-DWG03-00.</td>
<td>27/02/2023</td>
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<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SA00795SE</td>
<td>8K0CAB 8K0CB 70GCB (160HP) 70GCB (160HP) 70GCA 7ECA Installation of vortex generators on the wings and tail surfaces of the boeing 707-300 aircraft. The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended by additional or later amendments if indicated on FAA STC SA00795SE.</td>
<td>1. Drawing Package M2131, Revision A, dated 22 April 2001. 2. Installation Manual MA2132, Revision A, dated 22 April 2001. If more than five vortex generators are damaged or missing the aircraft is not airworthy. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>1. Drawing Package M2131, Revision A, dated 22 April 2001. 2. Installation Manual MA2132, Revision A, dated 22 April 2001. If more than five vortex generators are damaged or missing the aircraft is not airworthy. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/02/2023</td>
<td>Active</td>
<td>10081348</td>
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<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING SERVICES INC.</td>
<td>37-800</td>
<td>189 Passenger Seating Layout/This modification introduces a cabin layout using 63 economy class double seats generating a 150 passenger seating configuration. The Type and OSD Certification Bases for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval. The change is defined with revision 0 of AMES Engineering Order Document EO-0360001075_8M7H_08120.</td>
<td>- The change is defined with revision 0 of AMES Engineering Order Document EO-0360001075_8M7H_08120.</td>
<td>- The change is defined with revision 0 of AMES Engineering Order Document EO-0360001075_8M7H_08120.</td>
<td>28/02/2023</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>PME ENTERPRISES INC.</td>
<td>FAA SA206-04CH</td>
<td>JETSTREAM 3100 SERIES</td>
<td>Air Ambulance Conversion on Jetstream 3100/3200 series</td>
<td>Flight Manual Supplement dated July 21st 2008</td>
<td>Maintenance Manual Supplement dated July 29th 2008 Master Data List: Report No.6 dated September 28th 1992</td>
<td>No carriage of oxygen onboard and oxygen system of the Air Ambulance conversion to be made inoperative. Prior to installation of this change/repair it must be determined that the interface relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/03/2023</td>
<td>Active</td>
<td>10081506</td>
</tr>
<tr>
<td>SKYTEC AEROSPACE GmbH</td>
<td>737-8, 737-800</td>
<td>64C / 24Y Passenger Seating Layout This modification introduces a cabin layout using 8 economy class triple seats and 32 business class double seats generating a 88 passenger seating configuration.</td>
<td>The change is defined with revision B of Skytec Master Equipment List. Document MBD-738C111-001(1) The aircraft shall be operated in accordance with revision A of Skytec Aircraft Flight Manual Supplement Flight Manual Supplement AMD-738C111-0001 Maintenance shall be accomplished in accordance with revision A of Skytec Instructions.</td>
<td>Prior to installation of this change/repair it must be determined that the interface relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/03/2023</td>
<td>Active</td>
<td>10081507</td>
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</tr>
<tr>
<td>EICHELSDORFER FLUGZEUG BAU</td>
<td>LS6-C, LS6-C18, LS6-18W</td>
<td>LS 6</td>
<td>Installation of a turbine engine for self-sustaining propulsion</td>
<td>Installation of the turbine engine EASA E.108 in accordance with Skytec document LS6 STCPB DL. The engine change/repair must be determined in accordance with revision A of Skytec Instructions.</td>
<td>Prior to installation of this change/repair it must be determined that the interface relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/03/2023</td>
<td>Active</td>
<td>10081517</td>
<td></td>
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<tr>
<td>AEROLITE AG</td>
<td>REV. 1</td>
<td>PC-24</td>
<td>Installation of EMS Equipment P/N 024-0303000-501</td>
<td>Installation of EMS Equipment P/N 024-0303000-501</td>
<td>In accordance with Aeronaut Master Data List: MBD-22-009 rev N dated 28 March 2023. Aircraft Flight Manual Supplement. ICAO document FMS-024-0303000-51 rev 0 dated 01 February 2023. Instructions for Continued Airworthiness: Operations Manual.</td>
<td>Prior to installation of this change/repair it must be determined that the interface relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/03/2023</td>
<td>Active</td>
<td>10081517</td>
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<tr>
<td>RUAG AG</td>
<td>AS 350 B3</td>
<td>Night Vision Imaging System for AS 350 B3 SN 4913 including Night Vision Goggles NVG, White Phosphor FENN NG80-02 250 Rotor Altitude Indicator Modification to cockpit lighting system</td>
<td>The Night Vision Imaging System for AS 350 B3 SN 4913 including Night Vision Goggles NVG, White Phosphor FENN NG80-02 250 Rotor Altitude Indicator Modification to cockpit lighting system</td>
<td>Rotorcraft modified by this design change employs Night Vision Imaging System lighting, commonly named NVIS Lighting, which has been certified to ensure the aircraft is compatible with Night Vision Goggles (NVG). For this reason, any deviation from the cockpit or cabin configurations specified in this document...</td>
<td>06/03/2023</td>
<td>Active</td>
<td>10081396</td>
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<tr>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AVCON INDUSTRIES, Inc.</td>
<td>Installation of Wing Tip Hard Point Provision</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List No. 05661/1000, Rev. IR, signed 5 May 2021,&lt;br&gt;later revisions of the above listed document(s) approved/accepted of EASA in accordance with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2023</td>
<td>Active</td>
<td>10081405</td>
<td></td>
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<tr>
<td>AVCON INDUSTRIES, Inc.</td>
<td>Installation of Underwing Hard Points</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Aerospace Design &amp; Compliance (1) Master Data List B350-MDL-0035, Revision R1&lt;.&lt;&gt;.&lt;&gt;. dated 17 May 2019,&lt;br&gt;(2) Instructions for Continued Airworthiness 248-811, Revision R, dated 15 May 2019,&lt;br&gt;later revisions of the above listed document(s) approved/accepted of EASA in accordance with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2023</td>
<td>Active</td>
<td>10081406</td>
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<tr>
<td>AVCON INDUSTRIES, Inc.</td>
<td>Increasing the Maximum Zero Fuel Weight (MZFW) from 13,000 to 13,400 pounds</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>1) Aerospace Design &amp; Compliance, LLC., Master Data List B350-MDL-0090, Revision R, dated 10 June 2021,&lt;br&gt;(2) AVCON Industries, Inc., Aircraft Flight Manual Supplement (AFMS)&lt;.&lt;&gt;. dated 10 June 2021,&lt;br&gt;later revisions of the above listed document(s) approved/accepted of EASA in accordance with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>This STC is applicable to all serial numbers for the M43 Model.</td>
<td>07/02/2023</td>
<td>Active</td>
<td>10081408</td>
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<td></td>
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<tr>
<td>THALES AVIONICS Inc.</td>
<td>Installation of the upgraded Thales Radio Altimeter Transceiver Units, ERT550</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25 and/or CS26:5.611(a). The requirements for environmental protection and the associated certified noise and/or emissions remain applicable.</td>
<td>Thales Avionics, Index List Drawing No. 295039, Revision A, dated 10 February 2023,&lt;br&gt;later revisions of the above listed document(s) approved/accepted of EASA in accordance with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2023</td>
<td>Active</td>
<td>10081410</td>
<td></td>
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</tr>
<tr>
<td>PMV ENGINEERING</td>
<td>Cabin humidification and dryer system installationRev1. A320-251N added to the applicability list</td>
<td>The Type and ODI Certification Base for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS 25-324.18, 25.301, 25.303, 25.305(a)(x)(y), 25.307(a), 25.308(a), 25.61, 25.601, 25.603, 25.605, 25.611, 2.</td>
<td>TextronAir Aviation Models: B300 &amp; B300C aircraft. Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/02/2023</td>
<td>Active</td>
<td>10081415</td>
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<tr>
<td>AVATION BUSINESS CONSULTANTS INTERNATIONAL S.R.L.</td>
<td>Installation of Remote Sensing System On PBM installation olomere sensing system: Laser (class 3B), Camera, and Remote Sensing Equipment</td>
<td>The Type and ODI Certification Base for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS-23 Amendment 4.23.21(x)(b), 23.23.3(x)(b), 23.505(b)(y), 23.29(k)(b), 23.301(a)(y), 23.303, 23.305(a)(y), 23.307(a)...</td>
<td>Top Master Document List ECS-00120-TMDC Rev 2, Instructions for Continued Airworthiness ECS-00120-ICA-501 Rev02, Instructions for Continued Airworthiness ECS-00120-ICA-502 Rev01, later revisions of the above listed document(s) approved/accepted of EASA in accordance with the Technical Implementation Procedures of EU-USA Bilateral Agreement.</td>
<td>13/06/2023</td>
<td>Active</td>
<td>10081425</td>
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</table>

**Note:** The applicability of this certificate is limited to the SN listed identified in the ECS-00120-TMDC document. As per ICAO documentation, "Calibration and changes to the humidiifier settings are only allowed in accordance with CCS Configuration/Chart. Any application for an update of the CCS Configuration..."
10081450
ICELANDAIR EHF
REV. 1
737-8
New seating layout with 160 total passengers using AVIO seats with 28/36-Inch IFE and ISPS system/airfly: 1. Editorial change.

The Type and OSD Certification Basis for the original product as amendedby the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: CS 25 Amdt: 18, 129 (a)(b), 301(a), 302, 303, 307(a), 365(b), 561(b)(c), 571, 601, 603 etc.

The Certification Basis for the original product remains applicable without certification approval, except where amended by additional or later amendments if indicated on FAA STC S11298DS. The requirements for environmental protection and the associated certificated noise and/or emissions levels at the...
<table>
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<tr>
<th>STC Holder</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
</tr>
</thead>
<tbody>
<tr>
<td>airplus engineering GmbH</td>
<td>DORNER 228-201 - 212</td>
<td>Installation of a special mission radome</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/appraisal. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Installed and maintained in accordance with MMS-2470-50000Rev 1 operated in accordance with POH-2470-60000 Rev 1 (&lt;i&gt;n&lt;/i&gt;) or later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>As defined in the associated technical documents. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change(s) and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/03/2023</td>
<td>Active</td>
<td>10081565</td>
</tr>
<tr>
<td>STENE AVIATION, INC.</td>
<td>FAA STC SA4302WE</td>
<td>Installation of leading edge &quot;CUFF&quot; on each wing</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Stene Aviation Master Drawing List Number SASP-1, Revision A, dated 20 November 2015; (&lt;i&gt;n&lt;/i&gt;) or later revisions of the above listed document(s) approved/accepted/under EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair and any other previously installed change(s) and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/03/2023</td>
<td>Active</td>
<td>10081568</td>
</tr>
<tr>
<td>HELI-ONE (NORMAY) AS</td>
<td>REV. 1</td>
<td>AS 365 N3</td>
<td>Electronic Flight Information System EFIS-8000 Installation (ref.modification AHSE-74000Rev 1): Installation of 3rd pilot tube used for standby instrument.</td>
<td>Rev 0 Certification Program ref. AHSE-74001 Revision E, dated 19/02/2023 Product Change Declaration ref. AHSE-74000 Revision A, dated 27/03/2023 Declaration of Compliance ref. AHSE-74004 Revision A, dated 27/03/2023 Fight Manual Supplement ref. HTtno.1374 dated 24/03/2023 Rev. 1: Certification Program ref.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change(s) and/ or repair will introduce no adverse effect upon the airworthiness of the product. At time of approval of this change to type certificate...</td>
<td>29/04/2024</td>
<td>Active</td>
<td>10081574</td>
</tr>
<tr>
<td>ELISEN &amp; ASSOCIATES INC.ELISEN &amp; ASSOCIATES INC.</td>
<td>TOCA-A- LSA19-171/D</td>
<td>FALCON 7X</td>
<td>Modification Data Summary MDS-3039 Revision A: Duncan Aviation AFM Supplement CERTDOC-013008 Revision A: Duncan Aviation ICA CERTDOC-01142 Revision A: Eliseon MSEL Supplement MEL-3039 Revision A: later revisions of the above listed document(s) approved/accepted/under EASA in accordance with...</td>
<td>Applicable to F7X MSN 113 Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change(s) and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/03/2023</td>
<td>Active</td>
<td>10081597</td>
<td></td>
</tr>
<tr>
<td>LEE AEROSPACE, INC.</td>
<td>FAA STC ST04221NY</td>
<td>Installation of LAI Heated Windscreens</td>
<td>The Certification Basis for the original product remains applicable to this certificate/appraisal, except where amended by later amendments or additions to the associated certified noise and/or emissions levels of the or...</td>
<td>1. Master Data List, LAI MEL-135, Revision A, dated 12 January 2023; 2. Instructions for Continued Airworthiness (ICA) LAI ICA-135, Revision B, dated 12 January 2023; 3. later revisions of the above listed document(s) approved/accepted/under EASA in accordance with the Technical Implementation Procedures...</td>
<td>This STC is applicable to all serial numbers for the Mako Learjet Inc. Models: 31, 31A &amp; 35A Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change(s) and/ or repair will introduce no adverse effect...</td>
<td>29/03/2023</td>
<td>Active</td>
<td>10081604</td>
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</table>

The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.

Avidyne Corporation Master Document List AVI/DE 157 Rev. 00 dated 2 May 2013.<1>-<2>
<1> Warnings for Continued Airworthiness (ICA), AVI/DE 163, Revision 00 dated 25 March 2013.<1>-<3>
<3> For later revisions of this STC refer to Hardware & Software Interrelationships between this STC and other previously installed STCs.

Aircraft must be configured with an existing S-TEC Manual Electric Transponder System and with a S-TEC autopilot as follows in order to accomplish this STC (Existing S-TEC Autopilot System 20 with 30 ALTS-TEC Part Number 01260 <1> or 01261 <1>(Existing S-TEC Autopilot System 30S-TEC Part Number: 12661 , 12...)

20/03/2023 Active 10231620

10081613  AERODISA S.L.  A321-200/231/271NX  A320-251NX/252NX/-253NX  236 Passenger Seating Layout/This modification modifies a 236 passenger seating configuration type-pitching of the seat rows 1 through 4. 747-200B/-200C/-200F 747-300/-400/-400F NX A321-200/-251NX/-252NX/-253NX C33A 35-33/-A33/-B33/-C33/E33/E33A E33C/F33/F33A/F33C/P35/S35 V35/V35A/V35B Installation of Garmin GI 275 (Standby FAA STC Certification Basis

The Type and OSD Certification Bases (CB) for the original product remain applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certification/approval...

The change is defined with revision AS of Aerosida Master Data List AE02023-01M - MDK: The modification is defined with revision AS of Aerosida Service Bulletin Document AE02023-01M - MDK. For later revisions of the above listed document(s) approved/accepted under the EASA system...

Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.

31/03/2023 Active 10261613

10081614  JANZAIR CONSULTING SERVICES, INC.  FAA STO4431AT  747-300/400/400F  747-200B/-200C/-200F Installation of the Rockwell Collins TPR-901 Mode S transponders and Free Flight SBAS sensor for ADS-B Out capability

The Type and OSD Certification Bases (CB) for the original product remain applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certification/approval...

Master Data List 747-3999-1601 Rev C Instructions for Continued Airworthiness 747-3999-1604 Rev AIAircraft Flight Manual Supplement 747-3999-1610 Rev. Aor later revisions of the above listed document(s) approved/accepted under the EASA system in accordance with the Technical Implementation Procedures of...

As per ICA Document Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.

30/03/2023 Active 10261614

10081618  EIREMOTE AVIATION SERVICES LIMITED  A320-251N Installation of Slide-Raft on A320This change installs emergency slide-rafts instead of the emergency slides on the passenger/crowd doors. The forward (FWD) and aft (AFT) slides are changed on both the left and right sides of the aircraft.

The Type and OSD Certification Bases (CB) for the original product remain applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certification/approval...

-COM supplement referenced CCOM/ 1676-001 issue 03 dated 24th March 2023- CCDS supplement referenced CCDS-1676-001 issue 02 dated 01st March 2023- AMM supplement referenced AMS1676-001 issue 02 dated 01st March 2023- MDL referenced MDL-1676-001 issue 01 dated 30th March 2023

Aeroplanes modified per this STC (i.e. with slide-rafts at D1<1>-<2>D4) are considered to be a “variant” to the A320 without the modification(s), with slides at D1<1>-<2>D4. Cabinet crew operating aereoplanes modified per this STC must complete aircraft difference training in accordance with ORAC CC.

30/03/2023 Active 10261618

10081627  DASSAULT AIRCRAFT SERVICES (DAS)  FALCON 200EXE Installation of Thrust Reverser on Falcon 200EXE aircraft.Revision 1 upgrades the software to Version 3.20, reviews mechanical installation, and revises the system configuration for the installation of Thrust Reverser on Falcon 200EXE aircraft.Revision 1:

The Certification Basis for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.

Dassault Aircraft Services (1) Master Data List F2XW- 0010-005, Rev. (B), dated 6 June 2024<1>-<2>
<1> Warnings for Continued Airworthiness (ICA) F2XW-0010-155, Rev. (B), dated 24 June 2020<1>-<2>
<2> Airplane Flight Manual Supplement (AFMS) F2XW-0010-150 Rev. (B)<1>-<2> for later...

This alteration is applicable to Falcon 200EXE aircraft serial numbers 28 and subsequent.<1>GI-275 and GDC-720 hardware, software and configuration must conform to the software and configuration matrix F2XW- 0010-250 Revision A for the installation of this change/repair...

28/06/2024 Active 10261627
### List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC no.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>AIRBUS HELICOPTERS DEUTSCHLAND GmbH</td>
<td>10081634</td>
<td>EC135 T2, EC135 T3, EC135 P3, EC135 T1, EC135 T2, EC135 P1, EC135 P2</td>
<td>Installation of upgraded transponder Garmin GTX 330 ES (Extended Squitter) in accordance with Compliance Report SK-MAC-255-CR-P-2318</td>
<td>For the Major Design Change regarding the installation of the upgraded transponder Garmin GTX 330 ES (Extended Squitter) subject to approval, the Certification Basis (CB) for the original product remains applicable. In addition, the applicant has shown compliance with CS-ACN/Issue 4, as applicable...</td>
<td>- rotorcraft flight manual supplement SK-MAC-255-SLP-B-2317/Issue A</td>
<td>Compliance report SK-MAC-255-CRP-B-2318 issue A</td>
<td>Limited to helicopters with CPDLS as CPDLS only applicable to helicopters without installed TCAS system. Prior to installation of this change/repair, it must be determined that the...</td>
<td>03/04/2023</td>
<td>Active</td>
<td>10081634</td>
</tr>
<tr>
<td>AIR PROJET</td>
<td>10081635</td>
<td>DR 400 NGL, DR 400 R8, DR 400/200 R, DR 400/120 A, DR 400/100, DR 400/120 DR 315, DR 340, DR 360, DR 380, DR 300/180 R, DR 300/125, DR 300/140, DR 300/108, DR 300, DR 253, DR 253 B, DR 250, DR 250 B, DR 250-B-160, DR 250-180, DR 221, DR 221 B, DR 220 AB, DR 220 B, DR 220, DR 220 A, DR 220</td>
<td>Engine monitoring parameters CGR-30P &amp;/or Fuel level Indicator FL20XD &amp;/or OES Fuel level Sender</td>
<td>The Certification Basis for the original product as amended by following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS 23 Amdt 4 §5 23.603(n); 23.773, 23.867, 23.901, 23.954, 23.955, 23.993, 23.1121, 23.1301, 23.1304, 23.1308(n); 23.1318(n); 23.1301, 23.1304...</td>
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<td>None</td>
<td>03/04/2023</td>
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<td>10081635</td>
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<tr>
<td>SECURAPLANE TECHNOLOGIES, Inc.</td>
<td>10081645</td>
<td>A321-271N, A321-271N, A321-272N, A321-272N, A321-272N, A321-272N, A321-272N, A321-272N, A321-272N, A321-272N, A321-272N</td>
<td>Replacement of Main Ship Battery</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certification/approval.</td>
<td>Standard Aero Master Data List 1035677, Rev. A, dated 7 November 2019; the latest version of the above listed document(s) approved/accepted under the EASA Technical Implementation Procedures of EU USA Bilateral Agreement; Prior to installation of this change/repair it must be determined that the...</td>
<td></td>
<td></td>
<td>11/05/2023</td>
<td>Active</td>
<td>10081645</td>
</tr>
<tr>
<td>HELI-ONE (NORWAY) AS</td>
<td>10081651</td>
<td>AS 365 N3</td>
<td>Eurovion 7 and MFD Installation (ref. modification AHSE-74250)</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certification/approval w...</td>
<td>Certification Program Ref. AHSE-74250 Revision B, dated 30/03/2023/30 Product Change Declaration Ref. AHSE-74250 Revision A, dated 31/03/2023</td>
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<td>05/04/2023</td>
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<td>Ref</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10081653</td>
<td>FOKKER SERVICES B.V.</td>
<td>737-8, 737-9</td>
<td>Introduction of ULD on B737 Max (8-9) The change introduces an Underwater Locator Device.</td>
<td>- Certification Program CPR-0467 issue 2 Certification Record Report.</td>
<td>CRP-0467 issue 1 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>03/04/2023</td>
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<td>10081656</td>
<td>BOEING IRELAND LIMITED</td>
<td>737-8</td>
<td>189 Passenger Cabin Reconfiguration/Change from a two class, 174 passenger, to a full economy 189 passenger/cabin interior configuration.</td>
<td>The Type and OSD Certification Bases (CB) for the original products are applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
<td>Change / Repair Approval.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>06/04/2023</td>
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<td>10081658</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Certification of Alternative Floor Panels/This modification certifies alternate cabin floor panels as an replacement on various Boeing aircraft.</td>
<td>The change is defined with Revision A of Master Data List document/MDL-ALLAC01-350. The aircraft shall be maintained in accordance with Revision B installation Manual document/IM-ALLAC01-3500 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>11/04/2023</td>
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<td>10081678</td>
<td>GFD GmbH</td>
<td>36A</td>
<td>35A</td>
<td>Mechanical Provisions for the installation of an antenna in the upper fuselage</td>
<td>GFD, Master Data List 22-052-MDL-01, Rev. 0.1, 04.04.2023GFD, Engineering Order, 22-052-EO-01, Rev. 0.0, 13.12.2022 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>12/04/2023</td>
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<td>10081678</td>
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<tr>
<td>10081686</td>
<td>ZIM AIRCRAFT SEATING GMBH</td>
<td>A350-941, A350-1041</td>
<td>BCI seats installed in A350 Air Canada Aircrews modification due to Material Obsolescence.A. Plastic raw material change. Obsolete material XHR525 needs to be replaced for spares - Kydex FST (equal to XHR566) for backrest - Kydex 665HH for kick panel and center console decorative parts.B. For bon...</td>
<td>BCI seats installed in A350 Air Canadi...</td>
<td>Master Document List 565706-MDL-001, Revision 0 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This STC is limited to the following aircraft SN - A350-941: MSN 305; 315; 325 and 433- A350-1041: MSN 065; 482 and 533Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>12/04/2023</td>
<td>Active</td>
<td>10081686</td>
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<tr>
<td>10081692</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>REV. 1</td>
<td>AS 350 B3 Airframe-Mounted Fuel Filter Installation, compatible with existing aircraft's crash resistant fuel system.Revision 1 an administrative re-issue to correct the address</td>
<td>The change is defined with Revision A of Master Data List document/MDL-ALLAC01-350. The aircraft shall be maintained in accordance with Revision B installation Manual document/IM-ALLAC01-3500 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Compliance Program, CP-0402, issue B-Master Drafting List, MDL-21-001, Issue 8.Flight Manual Supplement, FMS-AHCA-172, Rev 1. Instructions for Continued Airworthiness, ICA-AHCA-097 Rev 1.3.1 or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the...</td>
<td>The following are pre-requisite mods for the subject installation. Arril 2D Engine (OP-4302) Crashworthy Fuel Tank (OP-4605) Off Fuel System Crash Resistance compatible with Cargo Swing Installation(07/2003)Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>09/04/2024</td>
<td>Active</td>
<td>10081692</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10081735</td>
<td>SWISS AIR-AMBULANCE Ltd.</td>
<td>AS 350 B3</td>
<td>NVIS Configuration R980-00085.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>Master Document List R180-00063 MDL Revision 5Certificate Programme R980-00019 Revision BFight Manual Supplement R980-00034 Revision CService Bulletin R980-00036 Revision OCA Supplement R980-00035 Revision 6 or later revisions of the above listed document(s) approved/accepted under the EASA system...</td>
<td>Rotarycraft modified by this design change employs Night Vision Imaging System lighting, commonly named NVIS lighting, which has been certified... ensures the aircraft is compatible with Night Vision Goggles (NVG). For this reason, any deviation from the cockpit or cabin configurations specified in this de...</td>
<td>19/04/2023</td>
<td>Active</td>
<td>10081735</td>
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<tr>
<td>10081752</td>
<td>DOA21 LIMITED</td>
<td>BD-500-1A10</td>
<td>Cabin air exhaust muffler installation of a &quot;muffler&quot; to reduce this noise to a more comfortable level by installing a valve cap and sound damping system.</td>
<td>The Type and OSD Certification Bases plus CS 25.571 adopted 19 The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise datab...</td>
<td>Master Document List D21-MDL-003 Issue 3 Master Document List D21-AMMS-0005 Issue 2 An aerodynamic limitation Section oc OCA Supplement Or later revisions of the above listed document(s) approved/accepted under the EASA system...</td>
<td>Airbus Service Bulletin BD500-215006 – Environmental Control – Cooling – Introduction of a Water-Activated Check Valve with a Housing needs to be installed prior to the installation of this Muffler STC...</td>
<td>21/04/2023</td>
<td>Active</td>
<td>10081752</td>
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<tr>
<td>10081756</td>
<td>AAXICO TECHNICAL SERVICES INC.</td>
<td>FAA STC ST04522AT</td>
<td>Installation of Tatar Aerospace LED Replacement Navigation/Position/Lights</td>
<td>The Certification Bases for the original product as amended by the following paragraph(s) at a later amendment CS25 armd t: 25.1389, 25.1387, 25.1390, 25.1391, 25.1393, 25.1395, 25.1397, 25.15020CS 25 armd t: 25.1390, 25.1391, 25.1393, 25.1395, 25.1397, 25.15020CS 25 armd t: 25.1390, 25.1391, 25.1393, 25.1395, 25.1397, 25.15020CS The Certification Bases for the original product as amended by the following paragraph(s) at a later amendment CS25 armd t: 25.1389, 25.1387, 25.1390, 25.1391, 25.1393, 25.1395, 25.1397, 25.15020CS The Certification Bases for the original product as amended by the following paragraph(s) at a later amendment CS25 armed t: 25.1390, 25.1391, 25.1393, 25.1395, 25.1397, 25.15020CS...</td>
<td>Tatar Aerospace Master Data List MDL-028-01or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement...</td>
<td>Prior to installation of this change it must be determined that the... ...interrelationship between this change/repair and any otherwise installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and models approved...</td>
<td>08/05/2023</td>
<td>Active</td>
<td>10081756</td>
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<tr>
<td>10081772</td>
<td>AIRBUS S.A.S.</td>
<td>A320-214</td>
<td>Overwing Exit Reactivation – MOD A0242</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>Certification Summary Report SRRA-23001 Issue 2AAM Supplement MOD A0242 Issue March 2023or later revisions of the above listed document(s) approved/accepted under the EASA system...</td>
<td>Prior to installation of this change it must be determined that... ...interrelationship between this change/repair and any otherwise installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and models approved...</td>
<td>25/04/2023</td>
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<td>10081772</td>
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<td>10081798</td>
<td>DOA21 LIMITED</td>
<td>BD-500-1A10</td>
<td>Introduction of a cabin Interior Rear-Type corrected in the documentation list</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment CS25 armd t: 25.1389, 25.1387, 25.1390, 25.1391, 25.1393, 25.1395, 25.1397, 25.15020CS 25 armd t: 25.1390, 25.1391, 25.1393, 25.1395, 25.1397, 25.15020CS 25 armd t: 25.1390, 25.1391, 25.1393, 25.1395, 25.1397, 25.15020CS...</td>
<td>Master Document List D21-MLD-002 Issue 3AAM Supplement D21-AMMS-0001 Issue 7Maintenance Planning Document D21-MLDP-002 Issue 10AAM FDD crew training D21-FCDTRN-0001 Issue 1or later revisions of the above listed document(s) approved/accepted under the EASA system...</td>
<td>- Airworthiness limitations as per Maintenance Planning Document - Airlane configurations changed per this STC are considered to be a &quot;variant&quot; to the A220 (BD500-100-300) commercial configurations. Cabin crew operating airplaces changed due to this STC...</td>
<td>21/08/2023</td>
<td>Active</td>
<td>10081798</td>
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<tr>
<td>10081801</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Installation of photoluminescent floor proximity emergency escape path marking system This modification installs a photoluminescent floor path marking system in the cabin to assist occupants in identifying the next exit in case of dense smoke condition.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated/certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>Master Document List D21-MLD-002 Issue 3AAM Supplement D21-AMMS-0001 Issue 7Maintenance Planning Document D21-MLDP-002 Issue 10AAM FDD crew training D21-FCDTRN-0001 Issue 1or later revisions of the above listed document(s) approved/accepted under the EASA system...</td>
<td>- Airworthiness limitations as per Maintenance Planning Document - Airlane configurations changed per this STC are considered to be a &quot;variant&quot; to the A220 (BD500-100-300) commercial configurations. Cabin crew operating airplaces changed due to this STC...</td>
<td>Prior to installation of this change it must be determined that the... ...interrelationship between this change/repair and any otherwise installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types...</td>
<td>28/04/2023</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>ELLIOTT AVIATION TECHNICAL PRODUCTSDEVELOPMENT LLC</td>
<td>FAA STC ST03292CH</td>
<td>SEE FAA AML</td>
<td>802.11 a/b/g/n WiFi System</td>
<td>Activation of an 802.11 a/b/g/n WiFi System</td>
<td>- Elliott Aviation activation instructions listed on AML No. ST03292CH&lt;++;&gt; - Master Data List, Report No. MDL-3766-111 Revision A&lt;++;&gt; - FAA STC Airbus STC Solutions MOD AI0253 - Airbus STC Solutions MOD AI0253</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and models approved...</td>
<td>02/05/2023</td>
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<td>10081804</td>
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<tr>
<td>SELKIRK AVIATION Inc.</td>
<td>FAA STC SA01007SE</td>
<td>SEE FAA AML</td>
<td>Interior Fiber Glass Panels</td>
<td>Manufacture and Installation of Interior Fiber Glass Panels</td>
<td>- Saftek Aviation Master Drawing List SA-5800, Rev. B dated December 2004&lt;++;&gt; - FAA STC Airbus STC Solutions MOD AI0253</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product. This STC is valid for aircraft types and models approved...</td>
<td>02/05/2023</td>
<td>Active</td>
<td>10081809</td>
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<tr>
<td>INFLIGHT CANADA Inc.</td>
<td>TCCA SA22-90 ISSUE 1</td>
<td>737-8</td>
<td>USB Power System Installation</td>
<td>The modification installs an USB Power System which is a new subsystem providing the USB power to the occupants of Row 1 thru 13 over R2 inc. R3 Y/C passenger seats (3 x 3 arrangement) and to the flight crew in flight deck.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>Installation shall be performed in accordance with Inflight Canada / DECA Aviation Master Data List MDL-21014, Revision 3 The aircraft shall be operated in accordance with Inflight Canada / DECA Aviation Flight Manual Supplement FMS-21014, Revision 1 Maintenance shall be performed in accordance with Inflight Canada / DECA Aviation Master Data List MDL-21014, Revision 3 The aircraft shall be operated in accordance with Inflight Canada / DECA Aviation Flight Manual Supplement FMS-21014, Revision 1 Maintenance shall be performed in accordance with Inflight Canada / DECA Aviation Master Data List MDL-21014, Revision 3 The aircraft shall be operated in accordance with Inflight Canada / DECA Aviation Flight Manual Supplement FMS-21014, Revision 1 Maintenance shall be performed in accordance with Inflight Canada / DECA Aviation Master Data List MDL-21014, Revision 3 The aircraft shall be operated in accordance with Inflight Canada / DECA Aviation Flight Manual Supplement FMS-21014, Revision 1 Maintenance shall be performed in accordance with Inflight Canada / DECA Aviation Master Data List MDL-21014, Revision 3 The aircraft shall be operated in accordance with Inflight Canada / DECA Aviation Flight Manual Supplement FMS-21014, Revision 1 Maintenance shall be performed in accordance with Inflight Canada / DECA Aviation Master Data List MDL-21014, Revision 3 The aircraft shall be operated in accordance with Inflight Canada / DECA Aviation Flight Manual Supplement FMS-21014, Revision 1 Maintenance shall be performed in accordance with Inflight Canada / DECA Aviation Master Data List MDL-21014, Revision 3 The aircraft shall be operated in accordance with Inflight Canada / DECA Aviation Flight Manual Supplement FMS-21014, Revision 1 Maintenance shall be performed in accordance with Inflight Canada / DECA Aviation Master Data List MDL-21014, Revision 3 The aircraft shall be operated in accordance with Inflight Canada / DECA Aviation Flight Manual Supplement FMS-21014, Revision 1 Maintenance shall be performed in accordance with Inflight Canada / DECA Aviation Master Data List MDL-21014, Revision 3 The aircraft shall be operated in accordance with Inflight Canada / DECA Aviation Flight Manual Supplement FMS-21014, Revision 1 Maintenance shall be performed in accordance with Inflight Canada / DECA Aviation Master Data List MDL-21014, Revision 3 The aircraft shall be operated in accordance with Inflight Canada / DECA Aviation Flight Manual Supplement FMS-21014...</td>
<td>04/05/2023</td>
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<tr>
<td>PT21 SOLUTIONS SAS</td>
<td>REV. 1</td>
<td>A200-235N-271N-272N-273N A320-231-232-233-251N A320-211-212-214-215-216</td>
<td>Cabin Installation: 180 YC seats in cabin plus associated monumentreimvolves and OHB OHSC changes/Rev. 1: removal of -787YC Recaro 3510A Seats Installation 180YCU Recaro 3510A Seats</td>
<td>Cabin Installation: 180 YC seats in cabin plus associated monumentreimvolves and OHB OHSC changes/Rev. 1: removal of -787YC Recaro 3510A Seats Installation 180YCU Recaro 3510A Seats</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>Configuration covered by Revision 1 of the certificate/applications per 23012/MDL/001. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/01/2024</td>
<td>Active</td>
<td>10081831</td>
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<tr>
<td>AIRBUS S.A.S.</td>
<td>A350-941</td>
<td>Airbus STC Solutions MOD A00253 - INSTALL FTTING AND VIASAT OAE</td>
<td>Airbus STC Solutions MOD A00253 - INSTALL FTTING AND VIASAT OAE</td>
<td>Airbus STC Solutions MOD A00253 - INSTALL FTTING AND VIASAT OAE</td>
<td>Certification Summary Report reference SRS-A-218B Issue G dated 02/05/2023 Modification Approval Sheet MAS n° A00253 issue 1 dated 02/05/2023 Instructions for Continued Airworthiness reference ICA MOD A00253 rev B dated 27 January 2023 or later revisions of the above listed document(s) approved/accepted</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/05/2023</td>
<td>Active</td>
<td>10081843</td>
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<td>EASA STC no.</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10081857</td>
<td>AIRBUS HELICOPTERS JAPAN CO., LTD.</td>
<td>H150-B</td>
<td>10081895</td>
<td>Partial Modification of the Anti-collision Light Including a Modified Anti-collision Light Switch Enabling the Selection of ‘Both’ Red And/Or White Alternating Lighting in Addition to the Existing ‘White’, ‘Red’ and ‘Off’ Positions.</td>
<td>The Type and OSD Certification Bases (CB) for the Original Product Preconditions are Applicable to this Certification/ Approval. The Requirements for Environmental Protection and the Associated Certification/ Approval Requirements are Unchanged and Applicable to this Certification/ Approval.</td>
<td>Certification Record document No CP-SA-491 &quot;Certification Record Partial Modification of the ANTICOLLISION LIGHT&quot; Revision 0 of 30 November 2022</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/05/2023</td>
<td>Active</td>
<td>10081857</td>
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<tr>
<td>10081861</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1 FAA ST0433SAT-D</td>
<td>GIV, GV</td>
<td>Installation of Honeywell Display Unit Software Upgrade with SXM Weather on the Gulfstream G-IV, G-V Aircraft. Rev. 1 corrects the Associated Technical Documentation Section.</td>
<td>The Type and OSD Certification Bases (CB) for the Original Product Preconditions are Applicable to this Certification/ Approval. The Requirements for Environmental Protection and the Associated Certification/ Approval Requirements are Unchanged and Applicable to this Certification/ Approval.</td>
<td>CES1 2088006, Revision F titled &quot;INDEX LIST – Honeywell DU-S85 Software Upgrade with Optional SXM Weather&quot; - CE1 2088003 Rev A titled &quot;Instructions for Continued Airworthiness – Honeywell DU-S85 v1.13 v2.03 Software Upgrade w/ optional SXM Weather-G-IV&quot; - CES1 208A005 Rev B titled &quot;Instructions for C...&quot;</td>
<td>For G-IV/G-V aircraft previously equipped with an XM weather receiver mfg p/n 4092743-901, the XM weather receiver must be upgraded to mfg p/n 4092743-903. This upgrade to the SXM weather receiver prior to conformance with DU-885DL-8596 software upgrade. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/06/2023</td>
<td>Active</td>
<td>10081861</td>
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<tr>
<td>10081895</td>
<td>VIH AEROSPACE Inc.</td>
<td>S-92A</td>
<td>737-800</td>
<td>Search And Rescue (SAR) Grab Handles Installation. The SAR Grab Handles Installation Allows for the Installation of an Additional Top Grab Handle and an Outboard Grab Handle in Any Combination to Aid SAR Crews While Performing Rescues.</td>
<td>The Type and OSD Certification Bases (CB) for the Original Product Preconditions are Applicable to this Certification/ Approval. The Requirements for Environmental Protection and the Associated Certification/ Approval Requirements are Unchanged and Applicable to this Certification/ Approval.</td>
<td>Master Document List V792-1719-MLD Rev. B dated Aug 18, 2012; Flight Manual Supplement V792-1719-FMS Rev. B dated Oct 12, 2022; Instructions for Continued Airworthiness V792-1719-ICA Rev. A dated Aug 19, 2022; Operations Manual V792-1719-OM Rev. N/R dated Apr 28, 2020; Certification Plan CP20028 Rev. 2 d...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/05/2023</td>
<td>Active</td>
<td>10081895</td>
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<tr>
<td>10081915</td>
<td>PARC AVIATION ENGINEERING SERVICES LIMITE D</td>
<td>737-800</td>
<td>189 Passenger Seating Layout/ This modification introduces a cabin layout using 63 economy class thalassophile seats generating a 189 passenger seating configuration.</td>
<td>The Type and OSD Certification Bases (CB) for the Original Product Preconditions are Applicable to this Certification/ Approval. The Requirements for Environmental Protection and the Associated Certification/ Approval Requirements are Unchanged and Applicable to this Certification/ Approval.</td>
<td>The change is defined with revisions 0 of Pan Aviation Engineering Services Master Document List 737-25185-MLD Maintenance shall be accomplished in accordance with revision 0 of ParC Aviation Engineering Services Instructions for Continuous Airworthiness Document 737-25185-ICA-01r for later revisions of the...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/05/2023</td>
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<td>10081919</td>
<td>DOA AESGLOBAL LIMITED</td>
<td>A321-232 A321-231 A321-213 A321-212 A321-211</td>
<td>737-800</td>
<td>215 Passenger Seating Layout/ This modification introduces a cabin layout using economy class seating generating a 215 passenger seating configuration.</td>
<td>The Type and OSD Certification Bases (CB) for the Original Product as Amended to the Following Additional or Alternative Airworthiness Requirements: The Following Paragraph(s) at a Later Amendment: CS 25.859 (A), 25.1389 (A), 25.1395 (A), 25.1393 (A), 25.1385 (A), 25.1383 (A), 25.1381 (A), 25.1379 (A), 25.1376 (A), 25.1374 (A), 25.1372 (A), 25.1369 (A), 25.1367 (A), 25.1365 (A), 25.1363 (A), 25.1361 (A), 25.1359 (A), 25.1357 (A), 25.1355 (A), 25.1353 (A), 25.1351 (A), 25.1349 (A), 25.1347 (A), 25.1345 (A), 25.1343 (A)</td>
<td>The change is defined with issues 1 of AESGLOBAL Master Data List AES-321-7246. Maintenance shall be accomplished in accordance with issues 1 of AESGLOBAL Instructions for Continuous Airworthiness Document AES-TR-3606 or later revisions of the above listed documents approved/ accepted/under the EASA system...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>Limitations/Conditions</td>
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<td>10081921</td>
<td>AIRBUS S.A.S.</td>
<td>STC Holder</td>
<td>A650/041</td>
<td>Airbus STC Solutions MOD A2025 - REMOVE GOOQ OAE AND INSTALL VASAT OAE/Remove GoO Antenna and Install VASAT OAE.</td>
<td>The Type and OSD Certification Bases for the original product as amended the following additional or alternative airworthiness requirements: Equivalent Safety Finding(s): ES/F B-302 - Verification/ validation compliance/certificate for large external antenna installation.</td>
<td>Prior to installation of this change/repair it must be determined the change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/05/2023</td>
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<td>10081921</td>
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<tr>
<td>10081927</td>
<td>AEROSPACE DESIGN &amp; COMPLIANCE LLC</td>
<td>FAA</td>
<td>B300, B300C</td>
<td>Installation of Becker Avionics DIVCS-6100 Digital Audio Management System.</td>
<td>The Type and OSD Certification Bases for the original product as amended the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: 23.1351(a)(1)(2) amndt 23-49, - 23.1353(h) amndt 23-49/25/23.1356(c) amndt 23-49/ - 23.1431(x)(a) amndt 23-43...</td>
<td>Master Data List B350- MDL-0505, Rev IR, dated 14 Feb 2020/Aircraft Flight Manual Supplement (AFMS) B350-AMFMS-0505, Rev A, dated 17 Apr 2020/Instructions for Continued Airworthiness (ICA) B350-ICA-0505, Rev IR, dated 06 Feb 2020/or later revisions of the above listed document(s) approved/accepted on behalf...</td>
<td>12/05/2023</td>
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<tr>
<td>10081942</td>
<td>AERODATA AG</td>
<td>FAA</td>
<td>B300, B300C</td>
<td>Installation of AD-ARIS-1210. Main items: Integration of an electrical load shed for FIS-power supply. Installation of still antennas mains on the pressurized fuselage - Integration of one operator console and one equipment console in the cabin - Integration of additional annunciators/...</td>
<td>The Type and OSD Certification Bases for the original product as amended to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Master Document List MDL-ARC-966, Rev 2/Aircraft Flight Manual Supplement AFMS-184, Rev 0/Aircraft Maintenance Manual Supplement AMM-1191, Rev 0/Minimum Equipment List MMEL-1015, Rev 0/or later revisions of the above listed document(s) approved/ accepted on behalf of the EASA syst...</td>
<td>16/05/2023</td>
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<tr>
<td>10081943</td>
<td>MAGNETIC MRO A5</td>
<td>FAA</td>
<td>ERU 190-200 STD</td>
<td>122 Passenger Seating Layout/This modification introduces a cabin layout generating a 122 passenger seating configuration including the installation of a In-Seat PowerSupply System.</td>
<td>The Type and OSD Certification Bases for the original product as amended to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>The change is defined with issue 2 of Magnetic MRO Master Data List/MMRO-C-0207/MDL-01/Maintenance shall be in accordance with issue 1 of Magnetic MRO AMM Supplement Document MMRO-C-0207-AMM-01/or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>16/05/2023</td>
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<td>10081947</td>
<td>ST ENGINEERING / AEROSPACE LTD.</td>
<td>FAA</td>
<td>A321-271N/-271NX/-272N</td>
<td>Installation of Lavatory for passengers with reduced mobility (PRM/Lavatory) in A320/A321 aircraft/Rev. 1</td>
<td>The Type and OSD Certification Bases for the original product as amended to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Prior to installation of this change/repair it must be determined the change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/05/2023</td>
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<td>1008151</td>
<td>HELI-ONE (NORWAY) AS</td>
<td>AS 332 L1 AS 332 L</td>
<td>Wall Mounted Seats (mod ref. AHSE-72700)</td>
<td>The Type and OSD Certification Bases (CB) for the original product precludes amending this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Program ref. AHSE-72702 Revision E, dated 31/03/2023; Product Change Declaration ref. AHSE-72700 Revision C, dated 31/03/2023; Declaration of Compliance ref. AHSE-72701 Revision C, dated 31/03/2023; FAA STC 10065111 must be previously or concurrently installed. Prior to installation of this change/repair it may be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>FAA Approved on 20 Sep...</td>
<td>17/05/2023</td>
<td>Active</td>
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<tr>
<td>10082016</td>
<td>AVIATION SPECIALITIES UNLIMITED, INC</td>
<td>REV. 1 FAA SR02600SE</td>
<td>429</td>
<td>Installation of a NVG Compatible Lighting and Filtration System Revision 1 is to remove the original limitations and update the STC with applicable references.</td>
<td>The Type and OSD Certification Bases (CB) for the original product precludes amending this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List (MDL), MDL-429004-3, Revision E, dated Sept 21, 2023; Instructions for Continued Airworthiness, ICA-429004-4, Revision IR, dated January 1, 2016, including Aircraft Specific ICA Appendices forSN 5707 dated May 5, 2023; RPM Supplement, FMS-429004-6, Revision A, FAA-approved on 20 Sep...</td>
<td>1. This supplemental type certificate applies to the model listed above only to the specific serial numbers as listed on the ASU MDL 2. Rotorcraft modified by this design change employs Night Vision Imaging System lighting, commonly named NVIS lighting, which has been certified to ensure the airworthiness of the aircraft.</td>
<td>25/09/2023</td>
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<td>10082016</td>
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<tr>
<td>10082017</td>
<td>LUFTHANSA TECHNIX AG</td>
<td>A321-231</td>
<td>EWG A321 Cabin modification/Increase of passenger seating capacity from 200 to 226 passenger seats</td>
<td>The Type and OSD Certification Bases (CB) for the original product precludes amending this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplemental Type Design Document STD2022-2210M Issue 00 Rev. 00: Instructions for Continued Airworthiness A-09-2038/ICA Issue 01 Rev.00; later revisions of the above listed document(s) approved.</td>
<td>FAASTC 10065111 must be previously or concurrently installed. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>31/05/2023</td>
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<td>10082018</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>REV. 1 FAA STC ST02740SE</td>
<td>777-300ER, 777-200</td>
<td>Interior Reconfiguration Rev. 1: adds Boeing 777-200 to the applicability list.</td>
<td>The Certification Basis for the original product as amendedby the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: C25/Issue 1: 25.613, 25.621, 25.791(a)(x)(v),( iv), 25.853(a)(x)(c)(iv), 25.256(c)(x)(c), 25.306(c)(iv)(b)(ii) Appendix F Part (The requirements for enviro...</td>
<td>Concurrent installation of Data Engineering EASA STC 10046965 Installation of a Panasonic In-Flight Entertainment System is required all aircraft modified by this STC. EASA STC 10046965 Installation of a Panasonic In-Flight Entertainment System is...</td>
<td>28/06/2024</td>
<td>Active</td>
<td>10082018</td>
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<tr>
<td>10082020</td>
<td>UUDS AERO CONCEPT</td>
<td>SEE EASA APPROVED MODELIST</td>
<td>Oxygen Bottle Installation</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amendedby the following additional or alternative airworthiness requirements: - Equivalent Safety Finding(s): F02/60083304 Maximum Oxygen Working Pressure and Pressure Limiting Performance Issue 2. - The following paragraph(s) at a later...</td>
<td>See EASA Approved Model List(s)(n) or later revisions of the above listed document(s) approved.</td>
<td>UUDS MOD UAC22M004 is a pre-requisite for this installation/installation is limited to configurations as per below Installation Instructions and Limitations Drawings UAC22M004-M0002, UAC22M004-M0003, UAC22M004-M0004, UAC22M004-M0005, UAC22M004-M0006, UAC22M004-M0007, UAC22M004-M0008, UAC22M004-M0009, UAC22M004-M0010...</td>
<td>31/05/2023</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>EASA STC No.</th>
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<td>10082027</td>
<td>AERODATA AG</td>
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<td>B300</td>
<td>Installation of Aerodata Flight Inspection System (FIS) AD-AFIS-113: Main items - Integration of an electrical load switch for FIS - Power supply - Installation of new antennas for FIS operation - Retraction of one GPS antenna - Integration of one operator console and one equipment rack in the cabin.</td>
<td>Master Document List MDL-AC-1-94. Rev. 2</td>
<td>Aircraft Maintenance Manual Supplement AFM-180. Rev. 2</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Applicable to Textron Aviation B300 aircraft, S/N FL-1158</td>
<td>01/06/2023</td>
<td>Active</td>
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<td>10082029</td>
<td>N AIR AVIATION INC</td>
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<td>PA-31</td>
<td>Sliding door, step, hand holds and restraint installation for sport/parachuting activities. This modification kit installs the following elements - a sliding door - an external hard hold - an external jumpstep - a restraint system for each parachute.</td>
<td>Master Data List AvSo-2205-MDL-1000 (Rev A) dated JANUARY 21, 2023</td>
<td>Flight Manual Supplement AvSo-2205-AFM-PA21-1000 Rev A dated January 21, 2023</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Serial Number 31-393 The aircraft must be equipped with FAA STCs SA00037SE and SA0700036 Prior to installation of this change/repair it must be determined whether interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the...</td>
<td>01/06/2023</td>
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<td>10082037</td>
<td>airbus engineering GmbH</td>
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<td>C90A C90GT C90GTi B300, B200, B200C, B200GT, B200CGT</td>
<td>Installation of Hensoldt Light-Weight Crash Recorder LCR 100</td>
<td>Master Drawing List ref. MDX-2436-310000 rev. 2</td>
<td>Engineering/Work Order Ref. EWO-2436-310000 Rev 1</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
<td>02/06/2023</td>
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<td>10082053</td>
<td>SOCIETE AIR FRANCE S.A.</td>
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<td>777-200</td>
<td>8777-200-287 Cable reconfiguration and IFE Panasonic &quot;AVI&quot; System rearrangement</td>
<td>CAS-B777-21-062 Change Approval Sheet &quot;Rev D&quot; (&lt;</td>
<td>&gt;) for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>- Aeronauts modified per this STC (i.e. with New IFE FIA Control Panel) are considered to be the same aircraft as the B777-200 with existing IFE/Panasonic &quot;AVI&quot; system (EASA STC 1046995, FAA STC 570317(IV)). Cabineers operating aeroplanes modified per this STC must complete aircraft training...</td>
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<td>10082080</td>
<td>MT-PROPELLER ENTWICKLUNG GMBH</td>
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<td>PA-28RT-201T (TURBO ARROW IV) PA-28R-201T (TURBO ARROW III)</td>
<td>Installation of MTV-12-D-188-53.</td>
<td>To be operated in accordance with MT-Propeller AFM-M Document No. E-40207 Rev. 0. According to MT-Propeller Master Documentation List No. E-418 Rev. 0 to be installed i.e. MT-Propeller Installation Instruction Doc. No. E-4201 Rev. 0 to be maintained i.e. MT-Propeller ICA Doc. No. E-4202 Rev. 0 or later...</td>
<td>- According limitation section of AFM-M Prior to installation of this change/repair it must be determined whether interaction between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>APTOZ EHF</td>
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<td>Navigation</td>
<td>New THALES ERT550R Radio Altimeter Installation</td>
<td>Master Data List MDL341431 (Reduction of Environmental Impact)</td>
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<td>07/06/2023</td>
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<td>BOEING IRELAND LIMITED</td>
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<td>LATAM 787-9</td>
<td>Cabin Reconfiguration/Revision 1 revises LATAM 787-9 Cabin Crew Seating Locations</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Boeing Global Service (BGS) TC Service Bulletins B787-81205-B8250283-00and B787-81205-B8250296-00, to modify the store 3ftm from a Type 'A' to a Type 'C' exit, must be implemented prior to the work carried out by BIL under this change/repair.</td>
<td>12/06/2023</td>
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<td>METAL INNOVATIONS INCORPORATED</td>
<td>FAA STC SA02763SE</td>
<td>Installation of Metal Innovations Inc. C-208B Cargo Pod Alteration</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval, except where amended or later amendments are indicated on FAA STC SA02763SE. The requirements for environmental protection and the associated certified noise and emissions levels of the product remain applicable to this certificate/approval.</td>
<td>1. MILDoc00307, Rev. B, dated 21 November 2022; 2. Master Drawing List M5D00305, Rev. C, dated 5 February 2023; 3. Instructions for Continued Airworthiness (ICA) dated 1 August 2022; 4. Aircraft Flight Manual Supplement (AFLM) M5D00189, Rev A, dated 3 May 2022 or later.</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/06/2023</td>
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<td>CHICAGO JET GROUP</td>
<td>FAA ST04188NW</td>
<td>GULFSTREAM 200/208</td>
<td>Installation of the Protrim Universal DL-150 Iodrum Transceiver to include LPV, TANS-1A, ATM-81 on Gulfstream 200 Aircraft</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Chicago Jet Group Master Data List C2001MDL00, Rev. 1, dated 06/10/2021, or later FAA Approved Revision Instructions for Continued Airworthiness (ICA) dated 06/10/2021, or later ICA or FAA approved revision.</td>
<td>06/06/2023</td>
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<td>ENVOY AEROSPACE, LLC</td>
<td>FAA STC ST04286CH</td>
<td>757-200/208/209</td>
<td>Retrofit Installation of Honeywell ALC-52B Radio Altimeter System</td>
<td>Prior to installation of this change/repair it must be determined whether the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Retrofit Installation of Honeywell ALC-52B Radio Altimeter System in accordance with Certification Master Data List, Doc. No. CN07-008MDL-00, Revision 5, dated 7 April 2023 or later.</td>
<td>24/07/2023</td>
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<td>AEROMETALS INC.</td>
<td>360H3</td>
<td>Extended Landing Gear Skid Assembly installation of FAA PMA Part Number 360H92114 Extended Landing Gear Skid Assembly replacement article.</td>
<td>The Type and OSD Certification for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Aerometals Instructions for Continued Airworthiness, 360H92114 Extended Landing Gear Skid Assembly, AMI-15, Rev. - 1st September 1998, Aerometals Report No. 009199, Revision - 16th June 1999, including Revision A 20th April 2000 or later revisions of the above listed document(s) approved/ accepted (if applicable).</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/06/2023</td>
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<td>ELLIOTT AVIATION TECHNICAL PRODUCTS LOMPENT, LLC</td>
<td>FAA STC SA03316CH</td>
<td>Activation of the 802.11 a/b/g/11/ac WiFi System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Elliott Aviation activation instructions listed on AML No. SA03316CH(+/-) &gt;Master Data List &gt;Flight Manual Supplementation FMS-3830-111 Rev.R &gt;Instructions for Continued Airworthiness ICA-3830-111 RevA, dated 7th February 2017(+) or later revisions of the above listed document(s) approved/ accepted (if applicable).</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/06/2023</td>
<td>Active</td>
<td>10082128</td>
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<tr>
<td>BOGGI AERONAUTICS S.R.I.L.</td>
<td>427</td>
<td>Installation of External Payload Mount System</td>
<td>The Type and OSD Certification for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Document List 427P00100-MDL-1 Issue 1 &gt;Flight Manual Supplementation 427P00100-FMS-1 Issue 1 &gt;Instructions for Continued Airworthiness 427P00100-ICA-10 Issue 1 for later revisions of the above listed document(s) approved/ accepted (if applicable).</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2023</td>
<td>Active</td>
<td>10082130</td>
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<tr>
<td>SABENA TECHNOICS BGC</td>
<td>A321-271NX, A321-272NX, A321-275NX, A321-251NX, A321-252NX, A321-253NX</td>
<td>Cabin Refurbishment with a cabin configuration from 240 Pax true to 202 Pax (190 Y/C &amp; 12 B/C) onboard two A321-253NX.</td>
<td>The Type and OSD Certification for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>As a pre-requisite requirement of this change/repair, completion of the above listed document(s) approved/ accepted (if applicable).</td>
<td>As a pre-requisite requirement of this change/repair, completion of the above listed document(s) approved/ accepted (if applicable).</td>
<td>23/06/2023</td>
<td>Active</td>
<td>10082147</td>
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<tr>
<td>AIRBUS DEFENCE AND SPACE S.A.</td>
<td>A330-243</td>
<td>Multi Role Transport Tanker (MRTT) UAT T2 (United Arab Emirates) Tranche 2(DCV0944-E and DCV0945-F)Rev. 1: Introduce the design change DCV1439-E enabling the use of the FlightDeck Rest Compartment (FDRC).</td>
<td>The Type and OSD Certification for the original product as amended by the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: CS 25.642 Initial issue for affected areas by this STC. This certificate/approval involves a change with an ...</td>
<td>“Conversion Definition A330 MRTT UAT T2 STC TE-FE-DT-200199 - Adated 22- MAR-2024 - Aircraft Flight Manual and Weight and Balance Manual – Aircraft Flight Manual Supplement (EASA)- Approved Model A330 MRTT UK – Tranche 2 - Aircraft Flight Manual TE-FE-DT-20056 - &quot;Limitations of Weight and Balance&quot;.</td>
<td>As a pre-requisite requirement of this change/repair, completion of the above listed document(s) approved/ accepted (if applicable).</td>
<td>26/06/2024</td>
<td>Active</td>
<td>10082156</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC/Order</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10082167</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>10082188</td>
<td>60</td>
<td></td>
<td>DUAL FMS UNS-1Ew Upgrade</td>
<td>The Type and OSD Certification Requirements for environmental protection and associated noise and emissions levels of the product are unchanged remain applicable to this certificate/approval w.</td>
<td>Certification report ref: L60X-QCM34-005 issued 14/6/2023; Flight Manual Supplement ref: L60X-QCM34-005-M1-APM, ADSG Rev 0 dated 23/05/2023, ICA-350004-04, revision 0 dated 23/05/2023</td>
<td>Active</td>
<td>100702167</td>
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<td>10082172</td>
<td>AIRWORK &amp; HEUSELEREIT GmbH</td>
<td>429</td>
<td></td>
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<td>Y-ropes for double load hook – External Cargo</td>
<td>The Type and OSD Certification Requirements for environmental protection and associated noise and emissions levels of the product are unchanged remain applicable to this certificate/approval w.</td>
<td>Pre-requisite for this installation: Dual External Cargo Hook Kit P/N 429-706-081 (supplier: BELL Textron), External mirror installation Kit P/N BHP 256 000 (supplier: BELL Textron). Prior to installation of this change/repair it must be determined that the inter-relationship between this change/repair...</td>
<td>Active</td>
<td>10082172</td>
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<tr>
<td>10082188</td>
<td>AVIATION SPECIALTIES UNLIMITED, INC.</td>
<td>FAA SR015905E</td>
<td></td>
<td>EC 130 B4, EC 130 T2, AS 350 B, AS 350 B3, AS 350 BA</td>
<td>Installation of a Supplemental Lighting system for NVIS operations.</td>
<td>The Type and OSD Certification Requirements for environmental protection and associated noise and emissions levels of the product are unchanged remain applicable to this certificate/approval w.</td>
<td>Rotorcraft modified by this design change employs Night Vision Imaging System (NVIS), which has been certified to ensure the aircraft is compatible with Night Vision Goggles (NVG). For this reason, any deviation from the cockpit or cabin configurations specified in this...</td>
<td>Active</td>
<td>10082188</td>
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<tr>
<td>10082199</td>
<td>KEHN BOREK AIR LTD.</td>
<td>TCAC SAAD-58, ISSUE 3</td>
<td></td>
<td>DHC-6 SERIES 330, DHC-6 SERIES 300 SERIES 310, DHC-6 SERIES 200 SERIES 210, DHC-6 SERIES 100 SERIES 110, DHC-6 SERIES 1</td>
<td>Removal of Beta Backup System</td>
<td>The Type and OSD Certification Requirements for environmental protection and associated noise and emissions levels of the product are unchanged remain applicable to this certificate/approval w.</td>
<td>Applicable to Viking Air Limited DHC-6 SERIES 100, SERIES 200 and SERIES 300 equipped with PT6A-27 or PT6A-34 Engines. Prior to installation of this change/repair it must be determined that the inter-relationship between this change/repair and any other previously installed change and/or repair will in...</td>
<td>Active</td>
<td>10082199</td>
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<tr>
<td>10082203</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>REV. 2</td>
<td></td>
<td>H160-B</td>
<td>Installation of SARPN ED 410 EOS turret mobile parts with hand controls in cockpit and cabin. Rev. 1 introduces revised performance information in the RFM supplement. The restrictions associated with post-STC action item CA 00-04 averted following the completion of the action items Rev. 2 cont...</td>
<td>The Type and OSD Certification Requirements for environmental protection and associated noise and emissions levels of the product are unchanged remain applicable to this certificate/approval w.</td>
<td>Helicopters with this STC are installed to restricted to category B. Prior to installation of this change/repair it must be determined that the inter-relationship between this change/repair and any other previously installed change and/or repair will in...</td>
<td>Active</td>
<td>10082203</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>AEROTECH &amp; CONCEPT</td>
<td>8737NG Cabin Full Pax, Configuration Change reconfiguration from 188Y to 188Y PAX</td>
<td>MASTER DATA LIST Doc, AE23511MDL Rev. 0 Or later revisions of the above listed document(s) approved/accepted under the EASA system</td>
<td>B/E AEROSPACE, Inc. - F gastrointestinal Interior Reconfiguration onboard 777-300ER series aircraft must be installed prior to concurrence with this STC. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously...</td>
<td>26/06/2023</td>
<td>Active</td>
<td>10082250</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>737-8 Interior OutfittingThe change consists of a complete cabin interior outfitting of the Boeing 737-8 with maximum 19 passengers VIP and 2 cabin crew for private and commercial use</td>
<td>The change is defined with Revision C of Master Data Document/MDL- LB01-2500. The aircraft shall be operated in accordance with Revision B of Airplane Flight Manual Supplement document AFMS-LB01-2500. The aircraft shall be maintained in accordance with Revision B of Instrictions for Continued Airworthiness.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelation between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/06/2023</td>
<td>Active</td>
<td>10082267</td>
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<td>DOA AESGLOBAL LIMITED</td>
<td>188Y Passenger Seating LayoutThe modification introduces cabin layout using 52 triplex economy class seats generating a 188 passenger seating configuration.</td>
<td>The change is defined with Issue 1 of AESGLOBAL Master Data Document/AES-737-7415MDL Maintenance shall be accomplished in accordance with Issue 1 of AESGLOBAL Instructions for Continued Airworthiness Document AES-TR-3707 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/06/2023</td>
<td>Active</td>
<td>10082268</td>
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<tr>
<td>SIA27-AT, SIA26-T</td>
<td>Installation of Single/Dual Garmin GTX 3000 Installation of either single or dual Garmin GTX 3000 transponder(s) replacing the existing transponder(s) and using existing XPD antenna and XPRD antenna cables. After the exchange the GTX 3000 installations fulfill the European airspace requirements for ADS-B.</td>
<td>Master Document List ASR-2023-002-MDL-01-00, Rev. Off. Flight Manual Supplement (Single Garmin GTX 3000) ASR-2023-002-FMS-05-1-00, Rev. Off. Flight Manual Supplement (Dual Garmin GTX 3000) ASR-2023-002-FMS-05-2-00, Rev. Obstructions for Continued Airworthiness (ECA), ASR-2023-002-ICA-11-00. Rev. Obstructions</td>
<td>Some enhanced surveillance airborne parameters are not available in aditional format and cannot be provided on request. Refer to AFMS (Single Garmin GTX 3000), ASR-2023-002-FMS-05-1-00, Rev. (for AFMS) (Dual Garmin GTX 3000), ASR-2023-002-FMS-05-2-00. Rev. Prior to installation of this change/repair it...</td>
<td>07/06/2023</td>
<td>Active</td>
<td>10082209</td>
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<td>SIA27-TT, SIA27-CG, SIA27-PC</td>
<td>The Type and OBD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a lateramendment: CS-23, Amdt. 4, 23, 29, 01-1, 031, 035, 307, 337, 561, 601, 605, 607, 609, 611, 613, 619, 625, 771, 777...</td>
<td>Master Document List ASR-2023-002-MDL-01-00, Rev. Off. Flight Manual Supplement (Single Garmin GTX 3000) ASR-2023-002-FMS-05-1-00, Rev. Off. Flight Manual Supplement (Dual Garmin GTX 3000) ASR-2023-002-FMS-05-2-00, Rev. Obstructions for Continued Airworthiness (ECA), ASR-2023-002-ICA-11-00. Rev. Obstructions</td>
<td>Some enhanced surveillance airborne parameters are not available in aditional format and cannot be provided on request. Refer to AFMS (Single Garmin GTX 3000), ASR-2023-002-FMS-05-1-00, Rev. (for AFMS) (Dual Garmin GTX 3000), ASR-2023-002-FMS-05-2-00. Rev. Prior to installation of this change/repair it...</td>
<td>07/06/2023</td>
<td>Active</td>
<td>10082209</td>
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<td>SIA27-TC, SIA27-CC, SIA27-AC, SIA27-AT, SIA26B-T</td>
<td>Modification to business class suites for the business class suites for B737NG Cabin Full Pax - AEROTEC &amp; CONCEPT</td>
<td>The change is defined with Revision C of Master Data Document/MDL- LB01-2500. The aircraft shall be operated in accordance with Revision B of Airplane Flight Manual Supplement document AFMS-LB01-2500. The aircraft shall be maintained in accordance with Revision B of Instructions for Continued Airworthiness.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelation between this change/repair and any other previously...</td>
<td>22/06/2023</td>
<td>Active</td>
<td>10082207</td>
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<td>STC Holder</td>
<td>System</td>
<td>STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limited/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>LETECKE OPRAVOVE TRÉNÍCNÁ, a.s.</td>
<td>STC Holder</td>
<td>L 410 UVP-E20</td>
<td>Installation of ADS-B OUT</td>
<td>The Type and OSD Certification Bases (CB) for the original production variant are applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval wo...</td>
<td>Master Document List No. 21-LOTM-885-MDL Rev 5; AFM Supplement No. 1- E91 and ADS-B OUTMST-708 Rev 2; AFM Supplement No.2; XPDTR 2 GTX 330-ES Rev 2; AFM Supplement No.3; GTN650X NAV/COM GPSYSTEM Rev 1; AFM Supplement No.4; Armor Connect, UNIT/19 Initial issue; AFM Supplement No. 5; TCAO II Processor...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/06/2023</td>
<td>Active</td>
<td>17552259</td>
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<tr>
<td>AKADEMISCHE FLIEGERGRUPPE BERLIN</td>
<td>STC Holder</td>
<td>GROB G 103 &quot;TWIN II&quot;</td>
<td>Twin III nose boom</td>
<td>The Type and OSD Certification Bases (CB) for the original production variant are applicable to this certificate/approval.</td>
<td>ANHANG ZUM FLUGHANDBUCH - GROB G103 - FOR DEN BETRIEB MIT NASENMAST - REVISION 04, 05, 06, 2023; Technische Mitteilung AFS-250-02 vom 21.06.2003; later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Only applicable, if saddlebag is prepared for installation of airco according to SB 315-525/2 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>04/07/2023</td>
<td>Active</td>
<td>17552252</td>
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<tr>
<td>DASSAULT AVIATION BUSINESSSERVICES LTD</td>
<td>STC Holder</td>
<td>FALCON 900EX (F900EX)</td>
<td>TAG-M372 - Varnish rework/Rework of cabin interior wood items finished with varnish</td>
<td>The Type and OSD Certification Bases (CB) for the original production variant are applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval wo...</td>
<td>TAG-M372-MCDL Rev A dated 19 June 2023Rev C, &gt;10 later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>This STC constitutes an Alternative Method of Compliance (AMOC) to the action required by EASA AD 2014-0002, sect. “Required Action(s) and Compliance Time (s),” paragraph (1). The STC approval and AMOC is limited to aircraft serial number 34. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect...</td>
<td>04/07/2023</td>
<td>Active</td>
<td>17552240</td>
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<tr>
<td>TRIEB GmbH</td>
<td>STC Holder</td>
<td>EB2NGR / EB 28 EDITION</td>
<td>Winglet Upgrade EB26/ EB26GR</td>
<td>The Type and OSD Certification Bases (CB) for the original production variant are applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval witho...</td>
<td>Trieb GmbH Drawing Number 0027-5731-00-00 - Revision A dated 22.06.2023; &gt;10 later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>None.</td>
<td>06/07/2023</td>
<td>Active</td>
<td>17552237</td>
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<tr>
<td>DOA AESGLOBAL LIMITED</td>
<td>STC Holder</td>
<td>737-800</td>
<td>188Y Passenger Seating Layout/This modification introduces accul layout using 62 triple economy class seats generating a 188 passenger seating configuration.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: 25.1301 (a), 25.1309 (a)(2), 25.1591 (a), 25.1353 (a) CS 25.1431 (a) at Amendment 4, CS25.1.25.29 (a) (b), 25...</td>
<td>The change is defined with the EASA Manual DataLinkAES-737-7419MDL. Maintenance shall be accomplished in accordance with issue 1 of AESGLOBAL Instructions for Continuous Airworthiness Document AES-TR-37 oil later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this issue 2 of AESGLOBAL Master DataLinkAES-737-7419MDL. Maintenance shall be accomplished in accordance with issue 1 of AESGLOBAL Instructions for Continuous Airworthiness Document AES-TR-370 later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>07/07/2023</td>
<td>Active</td>
<td>17552247</td>
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<td>SCANDINAVIAN AVIONICS DESIGN A/S</td>
<td>STC Holder</td>
<td>A321-231 A320-232</td>
<td>MAIN BATTERIES REPLACEMENT WITH ALTERNATE TYPE</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: EASA CS25 AMdt 2: CS 25.1, CS 25.29, CS 25.301, CS 26.302, CS 25.303, CS25.305, CS 25.307, CS 25.341, CS 25.561, CS 25.565, CS 25.601, CS 25.605, CS...</td>
<td>Engineering Order 3000097-ED Issue 15 of Certification Documents 3000097 DOC, Issue 15Continued Airworthiness 3000097-CA Issue 14MM Supplement 3000097 MM Issue 1or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>10/07/2023</td>
<td>Active</td>
<td>17552242</td>
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<td>No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10082346</td>
<td>JET AVIATION AG</td>
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<td>A340-313</td>
<td>Passenger Layout Reconfiguration: The modification is very similar to previous modifications completed on similar ships which had transferred from Swiss to Edelweiss. These aircraft were modified under an Airbus approved SB. The intention is that this modification will follow a similar scope, design...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS-25 Amend 27.</td>
<td>- Master Data List JBA- M-0381MDL-001 Issue D- Instruction for Continued Airworthiness JBJL- M-0381ICA-001 Issue A(0) later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Commercial operation / Non-smoking aircraft Prior to installation of this change/repair it must be determined that the interface relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/07/2023</td>
<td>Active</td>
<td>10082346</td>
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<td>10082372</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>REV. 1</td>
<td></td>
<td>737-800</td>
<td>Cabin Interior Reconfiguration: This change removes the existing two class cabin (8 BC &amp; 156 EY) to asingle class (189 Y/C) configuration. Rev. 1: correction of applicant's address</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: JAR-25.26(4) (5) and(8)(8) at Change 13: The requirements for environmental protection and the associated certification noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval within...</td>
<td>-NSI-13655-01-MDL-01 Master Data List: Issue 10r later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This STC is applicable to B-737-800 aircraft with LOPA configuration. DPA-378-3358 Rev B Prior to installation of this change/repair it must be determined that the interface relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/07/2023</td>
<td>Active</td>
<td>10082372</td>
<td></td>
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<tr>
<td>10082375</td>
<td>PATRIA AVIATION OY</td>
<td></td>
<td></td>
<td>A8412 EP</td>
<td>Flightdeal DJMx Connection/Installation of an Flightdeal DJMx cellular and data solution available in 3G/4G/LTE/450 MHz cellular networks.</td>
<td>The Type and OSD Certification Bases for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval within...</td>
<td>-Drawing List D-412-08- CS-07 revision D dated 27 June 2023: Installation document D-412-08-ED-25 revision D dated 27 June 2023: Instructions for Continued Airworthiness D-412-08-ICA-02 Issue 2 dated 20 June 2023: Rotorcraft Flight Manual Supplement D-412-08- RFM-15 Issue 1 Revision dated 26 April 2023...</td>
<td>Prior to installation of this change/repair it must be determined that the interface relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/07/2023</td>
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<td>10082375</td>
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<tr>
<td>10082376</td>
<td>GDC ENGINEERING GmbH</td>
<td></td>
<td></td>
<td>A320-232</td>
<td>Cabin Refit from (8BC /156 EY) to (189 Y/C) Configuration. Installation of new Door 1 LH and RH part...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements. The following paragraph(s) at a later amendment: CS-25.562 Amend. 27: The requirements for environmental protection and the associated certification noise and/or emissions levels of the product remain unchanged and remain applicable to this certificate/approval within...</td>
<td>- Description Report &amp; LOI: A320F-1DE-250000 Issue C: Master Certification Document List: CON- A320F-MCDL Issue 01: Instructions for Continued Airworthiness Supplement: A320F-ICA-OPA02 Master Document Register: A320F-ADR-250000 Issue E: Airplane Flight Manual Supplement: 43160AR006 RDR later...</td>
<td>This STC is limited to aircraft M9N 2229, 2238, and 2428. Prior to installation of this change/repair it must be determined that the interface relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/07/2023</td>
<td>Active</td>
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<tr>
<td>10082381</td>
<td>NSE</td>
<td></td>
<td></td>
<td>A1095</td>
<td>A1095 Trekker -- BCT Installation. The Box of Communication and Tracking (BCT) unit is a communication and tracking equipment collecting, transmitting, and storing tracking data.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: the following CS-27 paragraphs at Amend. 8: CS-27.21, CS-27.29, CS-27.301, CS-27.303, CS-27.305, CS-27.361, CS-27...</td>
<td>Certification Plan, Document No. PCERT-00431600A Issue A dated 04/07/2023: Installation Instructions, Document No. MD-00431600A Issue C1 dated 04/07/2023: Rotorcraft Flight Manual Supplement, Document No. RFM-BCT-Trecker-EN Issue 01...</td>
<td>Prior to installation of this change/repair it must be determined that the interface relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/07/2023</td>
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<td>10082381</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Country</td>
<td>Rev.</td>
<td>Foreign/NAA STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10082382</td>
<td>NSI</td>
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<td></td>
<td>A109S</td>
<td>The BS2 installation is not certified for maintenance purposes. Prior to installation of this change/repair it must be determined that the installation of this change/repair meets the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Certification Part No. PCERT-0004477014A Issue A3 dated03/07/2023; STC Data List. Document No. SDX-0004477014A Issue A1 dated 04/07/2023; Installation Instructions, Document No. MD-0004477014A Issue B1 dated04/07/2023; Rockcraft Flight Manual Supplement, Document No. RFMS-B21-Trekker-EN Issue A1...</td>
<td>2315-BM10-BM10-MDL Rev IR2315-BM10-ICA Rev IR2315-BM10-MDL Rev B2 or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>12/07/2023</td>
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<td>10082384</td>
<td>L2 CONSULTING SERVICES Inc.</td>
<td></td>
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<td></td>
<td>BD-700-1A11</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following: The change is defined with Revision 2 of Parc Aviation Engineering Services Master Document List 320-25444-... MDL-Maintenance shall be accomplished in accordance with Revision 0 of Parc Aviation Engineering Services Instructions for Continued Airworthiness Document 320-25444-ICA-01 or later revisions of...</td>
<td>The change is defined with Revision 2 of Parc Aviation Engineering Services Master Document List 320-25444-... MDL-Maintenance shall be accomplished in accordance with Revision 0 of Parc Aviation Engineering Services Instructions for Continued Airworthiness Document 320-25444-ICA-01 or later revisions of...</td>
<td>Prior to installation of this change/repair it must be determined that the installation of this change/repair meets the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/07/2023</td>
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<td>10082388</td>
<td>PARC AVIATION ENGINEERING SERVICES LTD</td>
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<td></td>
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<td>A320-232</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following: the installation of this change/repair meets the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>The change is defined with Revision 2 of Parc Aviation Engineering Services Master Document List 320-25444-... MDL-Maintenance shall be accomplished in accordance with Revision 0 of Parc Aviation Engineering Services Instructions for Continued Airworthiness Document 320-25444-ICA-01 or later revisions of...</td>
<td>Prior to installation of this change/repair it must be determined that the installation of this change/repair meets the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/07/2023</td>
<td>Active</td>
<td>10082388</td>
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<td>10082393</td>
<td>ALAMO ENGINEERING GmbH</td>
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<td>FALCON 2000</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following: the installation of this change/repair meets the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>The change is defined with Revision 2 of Parc Aviation Engineering Services Master Document List 320-25444-... MDL-Maintenance shall be accomplished in accordance with Revision 0 of Parc Aviation Engineering Services Instructions for Continued Airworthiness Document 320-25444-ICA-01 or later revisions of...</td>
<td>Prior to installation of this change/repair it must be determined that the installation of this change/repair meets the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/07/2023</td>
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<td>10082408</td>
<td>AERODATA AG</td>
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<td>B300C</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following: the installation of this change/repair meets the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>The change is defined with Revision 2 of Parc Aviation Engineering Services Master Document List 320-25444-... MDL-Maintenance shall be accomplished in accordance with Revision 0 of Parc Aviation Engineering Services Instructions for Continued Airworthiness Document 320-25444-ICA-01 or later revisions of...</td>
<td>Prior to installation of this change/repair it must be determined that the installation of this change/repair meets the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/07/2023</td>
<td>Active</td>
<td>10082408</td>
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<td>STC Holder</td>
<td>Upgrade of Honeywell MCS-6000</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>AIRBUS HELICOPTERS ESPANA, S.A.</td>
<td>The Type and OSD Certification Bases (CB) for the original product and the associated certified noise emissions levels of the product are unchanged and remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>-Rotorcraft Flight Manual Supplement SFLM-H135FH-2023-01 Normal/Revision 01- Rotorcraft Flight Manual Supplement SFLM-H135FH-2023-01 Normal/Revision 01- Compliance Record Document 2237M-EC135PST3 Revision A-</td>
<td>Limited to helicopters with Helionix cockpit variant. Prior to installation of this change/repair and any previously installed change and/or repair, the installation must be determined if there is an interference between this change/repair and any previously installed change and/or repair and if necessary, the STC holder will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/07/2023</td>
<td>Active</td>
<td>10082415</td>
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<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>The change is defined with Revision F of AMAC Master Data List document 950L-IV03-440. The installation must be accomplished in accordance with Revision A of AMAC Engineering Order document EC-IV03-4400 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Limited to helicopters with Helionix cockpit variant. Prior to installation of this change/repair and any previously installed change and/or repair, the installation must be determined if there is an interference between this change/repair and any previously installed change and/or repair and if necessary, the STC holder will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/07/2023</td>
<td>Active</td>
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<tr>
<td>E.A.D AEROSPACE AIRWORTHINESS TAECLIPSE TECHNICS</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: The following paragraph(s) at a later amendment:</td>
<td>The requirements for environmental protection and the associated certified noise emissions levels of the product are unchanged and remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Limited to MCS SATCOM system connected to high gain antennas as determined by a later paragraph at a later amendment.</td>
<td>20/07/2023</td>
<td>Active</td>
<td>10082420</td>
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<td>TEAM J.A.S., Inc dba PMA DESIGN AND MANUFACTURING</td>
<td>The Certification Basis for the original product remains applicable. The Certification Basis for the original product remains applicable.</td>
<td>The Certification Basis for the original product remains applicable.</td>
<td>Prior to installation of this change/repair and any previously installed change and/or repair, the installation must be determined if there is an interference between this change/repair and any other change/repair and if necessary, the STC holder will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/07/2023</td>
<td>Active</td>
<td>10082438</td>
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<tr>
<td>AIRFILM CAMERA SYSTEMS, INC</td>
<td>The Certification Basis for the original product remains applicable. The Certification Basis for the original product remains applicable.</td>
<td>The Certification Basis for the original product remains applicable.</td>
<td>Prior to installation of this change/repair and any previously installed change and/or repair, the installation must be determined if there is an interference between this change/repair and any other change/repair and if necessary, the STC holder will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/07/2023</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

<table>
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<tr>
<th>STC Holder</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Limits/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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</thead>
<tbody>
<tr>
<td>GDC ENGINEERING GmbH</td>
<td>A320-232</td>
<td>Cabin reconfiguration BHC162EY</td>
<td>The Type and OSD Certification Bases (CB) for the original product produce deficiencies applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master certification document list CCN- A32U-01-MDCL, issue 2</td>
<td>25/07/2023</td>
<td>Active</td>
<td>10082454</td>
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<tr>
<td>DRF STIFTUNG LUFTRETTLUNG DEMMENJUTGEG AG</td>
<td>EC135 P3, EC135 T3</td>
<td>Installation of lightweight flight recorder Hensoldt Lightweight CrashRecorder SferRec LOR100 in accordance with Master Document ListMD-135-31-32-00001</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: C5-27 Amendment 10a for 27.1, 27.21, 27.29; 27.21; 27.29; 27.31; 27.303; 27.305; 27.37; 27.561; 27.6.</td>
<td>- Master Document List MD-135-31-32-00001 Revision 01-Compliance Checklist CC-135-31-32-00001 Revision 00-Supplement PM5-10- DRF-068 Revision 01-Instructions for Continued Airworthiness CA-135-31-32-00001 Revision 00-later revisions of the above listed document(s) approved/accepted or later revisions of the document(s) approved/accepted under FAA STC-H-102381</td>
<td>26/07/2023</td>
<td>Active</td>
<td>10082458</td>
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<tr>
<td>LIFEPORT, LLC</td>
<td>FAA STC SA0072LA</td>
<td>Installation of a Patient Loading Utility System (PLUS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Master Drawing List (MDL) 520-1200, Revision DE dated 20 July 2020.&lt;br&gt;Instructions for Continued Airworthiness (CA), Document EGR-505-12, Revision F dated 26 August 2020.&lt;br&gt;Airplane Flight Manual Supplement No. EGR-505-11, Revision F dated 15 September 2020.&lt;br&gt;or later revisions of the above listed document(s) approved/accepted or later revisions of the document(s) approved/accepted under FAA STC.</td>
<td>27/07/2023</td>
<td>Active</td>
<td>10082466</td>
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<tr>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>CN-235-300</td>
<td>STC-AT-152, CASA CN235 (SAR) Provision-The aircraft is prepared for the installation of the SAR mission system with Structural reinforcement for camera and antenna installation. Installation of SAR mission equipment including exchange of existing mission display, mission equipment power distribution and installation of a new RFI shield.</td>
<td>The Type and OSD Certification Bases (CB) for the original product produce deficiencies applicable to this certificate/approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Supplement to AMM, Doc. No. STC-152-G-AMM, Rev. 1, Installation Instructions, Doc. No. STC-152-G-1, Rev. 00-later revisions of the above listed document(s) approved/accepted under EASA system.</td>
<td>28/07/2023</td>
<td>Active</td>
<td>10082475</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>ERJ 170-200 STD</td>
<td>Mounting device and power supply connection for PIVOT EFB solutions on ERJ-170.-Configuration 1: installation with sliding rail. Configuration 2: installation without sliding rail.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements. Special Condition(s): D-137 &quot;Mitigation of flight deck fireignition from lithium batteries that are not part of the aircraft design&quot; issue 4.</td>
<td>- Master Data List, Ref. 10007-01-A-MDCL, dated 30 April 2024&lt;br&gt;Airplane Flight Manual Supplement, Ref. 10007-01-A-AMMS-R01 dated 16 July 2023.&lt;br&gt;Instructions for Continued Airworthiness, Ref. 10007-01-A-ICA-R00 dated 29 Feb 2024 Master Minimum Equipment List Supplement Document, Ref. 10007-01-A-AMELS-R00.&lt;br&gt;Limitations as per §3 of the Aircraft Flight Manual Supplement document followed above. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the aircraft.</td>
<td>03/06/2024</td>
<td>Active</td>
<td>10082476</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>10082477</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>A330-243, A330-223, A330-203, A330-202, A330-201</td>
<td></td>
<td>Cabin Reconfiguration to 18 B/C &amp; 247 Y/C seats.</td>
<td>The Type and OSD Certification Bases (CB) for the original production items are applicable to this certificate/approval. The requirements for environmental protection and the associated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>Master Data List of NS.14180-01-MDL-01 issue 1 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10082482</td>
<td>PARC AVIATION ENGINEERING SERVICESLIMITE D</td>
<td>737-8</td>
<td>189 Passenger Seating Layout</td>
<td>This modification introduces a cabin layout by installing 63 economy class triple seats generating a 189-passenger seating configuration including associated changes in the cabin to adapt to the new layout.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: CS 25.952 (c)(5) and (c)(6), 25.785 (h)(1) and (h)(2) at Amendment 11The requirements for environmental protection and CHARM II Gas Detection System...</td>
<td>The change is defined with Revision 0 of Parc Aviation Engineering Services Master Document List 737-25186-MDL-Maintenance shall be accomplished in accordance with Revision 0 of Parc Aviation Engineering Services Instructions for Continued Airworthiness Document 737-25186-ICA-01 or later revisions of...</td>
<td>28/07/2023</td>
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<tr>
<td>10082483</td>
<td>AVIATION SUPPORT GmbH</td>
<td>MD900/MD902, MD900</td>
<td>Installation of CHARM II Gas Detection System (Generation II)</td>
<td>The Type and OSD Certification Bases for the original product are applicable to this certificate/approval. The requirements for environmental protection and the associated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>-Projektbeschreibung, ASE-91408, CHARM 2 – Gas Detection System, MD900, Doc. No. P-ASE-91408, Rev. 1-4 dated 28.04.2023; Liste der gültigen Dokumente, ASE-91408, CHARM 2 – Gas Detection System, MD900, Doc. No. MDL-ASE-91408, Rev. 0-1 dated 18.07.2023; Flight Manual Supplement, Installation of CHARM II.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/07/2023</td>
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<td>10082501</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>695, 690D, 690C, 690A, 690</td>
<td>Installation of MTV-37-1.6-C-F-(R)(G)/CFRL228-352 (Propeller on Twin Commander Aircraft Models 690) and 695.</td>
<td>According to MT-Propeller Master Documentation List No. E-3883 Rev. 0 to be installed in accordance with MT-Propeller AFM-S Document No. E-3885Rev. 0 to be installed i.a.w. MT-Propeller Installation Instruction Doc. No.E-3888 Rev. 0 to be maintained i.a.w. MT-Propeller ICA Doc. No. E-3887 Rev. 0e later...</td>
<td>According limitation section of AFM-SP Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/07/2023</td>
<td>Active</td>
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<tr>
<td>10082502</td>
<td>DOA AEGLOBAL LIMITED</td>
<td>F28 MARK 0100</td>
<td>ACTIVATION OF BLUEBOX WIFI SYSTEM</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.899(a), CS 25.1361(a)(c), CS 25.1393(a)(b), CS 25.1531(a)(b)(c), CS 25.1431(a)(d)(i) atAmendment 4; CS 25.1, 25.601;...</td>
<td>Master Data List of AES-100-6609M1L, Issue 2;Instructions for Continued Airworthiness Report AES-TR-3559 Issue 1AES FLIGHT MANUAL SUPPLEMENT No. 104 ISSUE 0101 later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The Airplanes must be modified according to Minor Change AES-100-6609M1L Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product...</td>
<td>31/07/2023</td>
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<td>STC Holder</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>A321-253N-271N-272N, A321-213-311-321-325N, A322-111-112-131-211-212</td>
<td>The modification consists in the recalculation of the cabin from 220PAX to 212 PAX (4BC + 208 YC)</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: CS.25-25, Amendment 4 CS.25-1, 25-29, 25-301, 25-303, 25-305, 25-307, 25-321, 25-337, 25-341, 25-385, 25-471,...</td>
<td>Certification Part 41135-01-A-CP-R01 Master Data List 41135-01-A-MDL-R00/Instructions for Continued Airworthiness 41135-01-A-ICA-R90</td>
<td>As a pre requisite of this change, the installation of SABENA TECHNICS AGAC (3N7-BOO) SB 41060-01-A-SB (A321-Cabin Recomconfiguration from 195PAX (12 B/C+183 Y/C) to 220 PAX (202 YC) is needed. The technical content of the SB is approved under the authority of Design Organisation Approval No. EASA-211...</td>
<td>04/06/2023</td>
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<tr>
<td>AVIONIK ENTWICKLINGS GmbH</td>
<td>525, 525A, 526</td>
<td>Installation of SwV 6.72 Update for Single Garmin GTN 725/750 compatible installation of Single Garmin GTN 725(X) / GTN 750(X) unit and optional Garmin Flight Stream FS 5/10</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: CS.25-23, Amend. 4- 23, 29, 28, 251, 301, 303, 305, 307, 337, 386, 561, 571, 801, 603, 606, 607, 609, 611, 613, 625,...</td>
<td>- Master Documents List, ASP-2022-037-MDL-01-00, Amend.00, 13.06.2023- AFMS (Honeywell SP2-5000), ASP-2022-037- FMS-05-1-00, Amend. 06.12.2023- AFMS (Collins Pro Line 21), ASP-2022-037- FMS-05-2-00, Amend. 00.12.06.2023- Instructions for Continued Airworthiness, ASP-2022-037- ICA-15-00, Amend.00, 12.06.2023</td>
<td>Design and data documents already EASA approved with STC 10042753, Rev.0 remain valid if legacy Garmin GTN 725 / GTN 750 units are installed. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will not introduce any adverse effect upon the airworthiness of the product. This STC is only valid for small aeroplane types and models...</td>
<td>04/08/2023</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>STC Ref.</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<tr>
<td>10082546</td>
<td>MT-PROPRIETOR ENTWICKLUNG GmbH</td>
<td></td>
<td></td>
<td></td>
<td>210M, 210N, 210R 210H, 210J, 210K, 210L, 210D, 210E, 210P, 210G</td>
<td>Installation of MTV-9/0200-65K ( ) Propeller on Textron Aviation Inc.210( ) Series.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment:C8-36 Amrd 6: FAR 36 Amrd 36:31: This certificate/approval involves a change with an impact on the noise database..</td>
<td>According to MT-Proprietor Master Documentation List No. E-3940 Rev. 17 to be operated in accordance with MT-Proprietor AFM-SP No. E-3940 Rev. 17 to be installed i.a.w. MT-Proprietor Installation Instruction Doc. No. E-3946 Rev. 17 to be maintained i.a.w. MT-Proprietor ICA Doc. No. E-3947 Rev. 01 later...</td>
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<tr>
<td>10082548</td>
<td>E.A.D AEROSPACE AIRWORTHINESS TIAECLIPSE TECHNICS</td>
<td>REV. 1</td>
<td></td>
<td></td>
<td>737-8</td>
<td>IA-BAND SATCOM Installation on B737-8(WAXA)-BAND SATCOM installation is composed of large antenna installations L.RUs in Cabin. The modification consists in the installation with reactivation of installed equipment.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - Equivalent Safety Finding(s): CR1-001: Vibration/Buffeting compliance criteria for Large external Antennas installation. The requirements for environmental or...</td>
<td>- AFM supplement Ref: 01S77-EAS-D-AFM-01 Issue B dated 31 July 2023- Master Data List Ref: 01S77-EAS-D-MDL Issue B dated 03 August 2023- ICA supplement: 01S77-EAS-D-ICA-01 Issue B dated 28 July 2023 or later revision of the above listed document(s) approved/accepted under the EASA system..</td>
<td>The airplane must be operated in accordance with AFM supplement 01S77-EAS-D-AFM-01 Issue B and maintained in accordance with ICA supplement: 01S77-EAS-D-ICA-01 Issue B. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously/installed change and/or repair will introduce no adverse affect upon the airworthiness of the product...</td>
<td>08/06/2023</td>
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<tr>
<td>10082558</td>
<td>AIRFORCE TURBINE SERVICE LTD.dba AIRFORCE TURBINE SERVICE</td>
<td>FAA STC</td>
<td>SE77765W</td>
<td></td>
<td>PT6A-15AG, PT6A-27, PT6A-28</td>
<td>Modification of the Pratt &amp; Whitney Canada PT6A-27, PT6A-28or PT6A-15AGb (t) PT6A-34 configuration</td>
<td>The Certification Basis for the original product remains applicable without certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Airforce Turbine Service Report No.001, Revision 0, dated 28 December1988 ( ) ( ) or later revisions of the above listed document(s) approved/accepted under the EASA in accordance with the Technical Implementation Procedures of EUI USA Bilateral Agreement..</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously/installed change and/or repair will introduce no adverse affect upon the airworthiness of the product...</td>
<td>09/06/2023</td>
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<td>10082583</td>
<td>ARCHIDEON HOLDINGS, L.L.C.</td>
<td>FAA STC</td>
<td>SAA5494AT</td>
<td>SEE FAA AML</td>
<td>PT6A-15AG</td>
<td>Installation of an Avionica, LLC Quick Access Recorder/Data Concentrator/monitor avRDC MAX (GAR) PIN#04-1250-32 and optional Avionica, LLC Cellular Module LTE (avCM LTE)PIN S05-0100-05</td>
<td>The Certification Basis for the original product remains applicable without certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Master Data List 22l590-MDL, Revision A, dated 16 September2022 ( ) ( ) or later revisions of the above listed document(s) approved/accepted under the EASA in accordance with the Technical Implementation Procedures of EUI USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously/installed change and/or repair will introduce/cause an adverse affect upon the airworthiness of the product...</td>
<td>08/06/2023</td>
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<tr>
<td>10082584</td>
<td>AVIATION SPECIALTIES UNLIMITED, INC.</td>
<td>REV. 1</td>
<td>FAA S01476SE</td>
<td>A109S</td>
<td>NVG Supplemental Lighting and Filtration System Revision 1 is to remove the original limitations and update the STC without applicable references.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval...</td>
<td>Master Drawing List (MDL), MDL-119004-3, Revision CD, dated20 May 2024,Instruction for Continued Airworthiness, ICA-119004-4, Revision H dated 09 July 2020, including Aircraft Specific ICA Appendices for S/ N923119 dated 26 July 2023- ICA-Supplement, EASA- RMS-119004-6, Revision IR, dated 16 Apr 2024...</td>
<td>1. This supplemental type certificate applies to the A109S helicopter model equipped with the Trettier kit and only to the specific serial numbers listed in the ASU MDL.2. Rollcraf modified by this design change employs Night Vision Imaging System lighting, commonly named NVIS lighting, which has been...</td>
<td>03/06/2024</td>
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An agency of the European Union
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<th>EASA STC no.</th>
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<th>Aircraft Type</th>
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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>10082585</td>
<td>ENFLITE, LLC d/b/a ENFLITE</td>
<td>FAA STC ST10565C</td>
<td>BD-70-2412</td>
<td>Installation of an Enflite Model 42100 Pocket Door between the forwardgalley and the passenger compartment, without theelectric door optioninstalled.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. EASA STC no.</td>
<td>08/08/2023</td>
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| 10082589    | THALES AVIONICS Inc. | FAA STC ST02806LA | 737-900ER | Installation of the upgraded Thales Radio Attimeter Transceiver Units,ERTS | The Certification Basis for the original product as amendedby the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment:CS25.13.171(a)(1)(2):
The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. | 08/08/2023 | Active | 10082589 |
<p>| 10082590    | GULFSTREAM AEROSPACE CORPORATION | FAA STC ST04311AT-D | GULFSTREAM 200/737-600/737-700/737-800/737-900/737-900ER | Installation of a Dual Universal Avionics System Corporation UNS-1E Flight Management System | The Certification Basis for the original product as amendedby the following additional or alternative airworthiness requirements: Special Condition(s):- FAA Special Condition (SC) 25-667-SC:Gulfstream Aerospace LP, Model Gulfstream 200 Aircraft: Non-Rechargeable Lithium Battery Instal... | 09/08/2023 | Active | 10082590 |
| 10082593    | AEROSPACE DESIGN &amp; COMPLIANCE LLC | FAA STC ST04011H-Y | 5550 580 590 | Installation of ENode Engine Data Acquisition Unit and I0Node DigitalData Recorder | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Aerospace Design &amp; Compliance, LLC:(&lt;&gt;) (1) Master Data List(&lt;&gt;):&lt; C5550-MLD-0017, Revision IR(&lt;&gt;) dated 25 November 2020(&gt;)&lt; (2) Instructions for Continued Airworthiness (ICA):(C5550-ICA-0071, Revision IR, dated 13 November 2020(&lt;&gt;):&lt; (3) other later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Proced... | 10/08/2023 | Active | 10082593 |
| 10082598    | PROTOPLANE SPECIAL MISSION AIRCRAFT | STC Holder | 1900D | Attendant seat installation | The Type and OSD Certification Basis(CSB) for the original productremains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval. Drawing List, document PP-145-DL-01Supplement to Aircraft Flight Manual, document PP-145-APM-01Instructions for Continued Airworthiness, document PP-145-54-ICA-02(a) other later revisions of the above listed document(s) approved/accepted under the EASA system. Limited to Textron Beechcraft aircraft 1900D S/Ns US. The aircraft must be equipped with the oxygen mask PN 129-380405-3, its cover PN 129-380405-49 and its surrounding part PN 129-380405-73: the oxygen tubes PN 101-380409-13Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. Door finish requirements are determined by the install... | 10/08/2023 | Active | 10082598 |</p>
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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
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<th>STC ref.</th>
<th>Model(s)</th>
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<th>Associated Technical Documentation</th>
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<tr>
<td>10082596</td>
<td>TABY AIRMANTENANCIA &amp; AKTIEBOLAG</td>
<td>-</td>
<td>SAAB 2000</td>
<td>-</td>
<td>GPS Landing System and ADS-B out information</td>
<td>This modification provides precision approach (LPV) ILS Look alike guidance on the SAAB 2000 together with GPS information to existing Transponder System for ADS-B out information</td>
<td>-</td>
<td>-</td>
<td>The airplane must be operated in accordance with AFM supplement ref.: 321006-C-AM-3461-002-07 and maintained in accordance with ICA supplement ref: 321006-C-CA-3461-001-01 Issue 01 dated 22nd April 2022 Prior to installation of this change/repair it must be determined that the interrelationship...</td>
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<td>10082600</td>
<td>CAVOK</td>
<td>-</td>
<td>B300</td>
<td>-</td>
<td>GPS antenna installation/removal</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25 Amdt 2: 25.120.21; 25.26; 25.301; 25.303; 25.305; 25.307; 25.321; 25.333; 25.337; 25.341; 25.651; 25.601,...</td>
<td>-</td>
<td>-</td>
<td>Service Bulletin, document 350-115-EB-007 Issue B, dated 02/08/2023; Compliance Record, Document No. 3500/104-COR Issue 1 dated 27.07.2023; AFM Supplement, Document No. 3500/104-AM A Issue 1 dated 10.07.2023</td>
<td>10/08/2023</td>
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<td>10082608</td>
<td>BOEING IRELAND LIMITED</td>
<td>737-8</td>
<td>Two to one class 189 passenger</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25 Amdt 2: Paragraphs CS 29.1, CS 29.21, CS 29.25, CS 29.251, CS 29.301, CS 29.303, CS 29.305, CS 29.307, CS 29.309, CS 29.310,...</td>
<td>-</td>
<td>-</td>
<td>Drawing List, Document No. 3000104-DL Issue 1 dated 27.09.2022; Engineering Order and Certification Program, Document No. 3500/104-E3/E301 dated 08.08.2023; Compliance Record, Document No. 3500/104-COR Issue 1 dated 27.07.2023; AFM Supplement, Document No. 3500/104-AM A Issue 1 dated 10.07.2023</td>
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<td>10082610</td>
<td>ALPINE AEROTECH LP</td>
<td>TCCA SHO2-40</td>
<td>AS 355 F2, AS 355 N, AS 355 MP, AS 355 E, AS 355 F, AS 355 B1, AS 355 B, AS 355 B1, AS 355 B2, AS 355 B3, AS 355 BA, AS 360 D</td>
<td>Bear Paw Kit Installation</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>-</td>
<td>-</td>
<td>Change / Repair Approval, 737-8: Two to One Class 189PAX USB Powered switchover from a two class, 174 passenger cabin, to a full economy 189 passenger cabin</td>
<td>16/08/2023</td>
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<td>10082618</td>
<td>J &amp; C AEROSPACE DESIGN AND CERTIFICATI ON SERVICES EUU Ltd.</td>
<td>A319-115</td>
<td>New IFE installation &amp; CMS upgrade/Replacement of obsolete equipment of the CMS/IFE with updated Airborne equipment and installation of electrical provision for the cabinsecurity system.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>-</td>
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<td>MDL referenced JC2000-00900-001 dated 14 AUG 2023;(&gt;) or later versions of the above listed document(s) approved/accepted under the EASA system.</td>
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<td>FOKKER SERVICES B.V.</td>
<td>1008261</td>
<td>737-400ER 737-800 737-900 737-900 737-700 737-400</td>
<td></td>
<td>Activation of SBAS system on B737NG/737-7/8 with MOP/CP/CR/0472. Development</td>
<td>The Type and OSD Certification Bases for the original product and additional alternative airworthiness requirements.</td>
<td></td>
<td>This STC must be installed in conjunction with Fokker Services certification not approved for CR/</td>
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<td>FAA STC SA09796DS</td>
<td>PA-39</td>
<td>ST-01326</td>
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<td>Installation of metallic window blanks for Britten-Norman BN-2A-2, BN-2A-6, BN-2A-8, BN-2A-26, BN-2A-6</td>
<td>The Certification Basis for the original product remains applicable without certificate approval.</td>
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<td>FAA STC SA09796DS</td>
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<td>ST-01134</td>
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<td>Installation of metallic window blanks for Britten-Norman BN-2A-2, BN-2A-6, BN-2A-8, BN-2A-26, BN-2A-6</td>
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<td>Installation of metallic window blanks for Britten-Norman BN-2A-2, BN-2A-6, BN-2A-8, BN-2A-26, BN-2A-6</td>
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<td>Installation of metallic window blanks for Britten-Norman BN-2A-2, BN-2A-6, BN-2A-8, BN-2A-26, BN-2A-6</td>
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<td>10082654</td>
<td>CAVOK</td>
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<td>P180 ADI-8 OUT Installation with GTX 3000</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: 23.26, 23.29, 23.251/23.301 a) b) c) 23.302, 23.305 a) b) 23.307 a) 23.321 a) 23.341 a) 23.365 a) d) ..., DCM03 EN 15329-035; CFF-002, Issue R, AFMS no. 180-003-H002, Issue 1; C(AMIS) no. 190-003-H001, Issue 1 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>DCM03 EN 15329-035; CFF-002, Issue R, AFMS no. 180-003-H002, Issue 1; C(AMIS) no. 190-003-H001, Issue 1 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<td>10082658</td>
<td>DIGISKY S.R.L.</td>
<td>P2006T</td>
<td>Installation of LIDAR system</td>
<td>The Type and OSD Certification Bases (CB) for the original product/commercially applicable to this certificate/approval The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval w...</td>
<td>Master Data List Doc. TP3200-01-00-01M-01 issue 1, dated 1 July 2023; Aircraft Flight Manual Supplement Doc. TP3200-09-AFMS-01 issue 1 dated 31 July 2023 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>As a prerequisite for the installation of the Ladar system, the Tecnam P2006T aircraft must be equipped with: Big cabin hatch/mod (mod 2006/207), Convertor Box (mod 2006/204), Support frame (mod 2006/207) and Transav/LSG Antenna installed i.a.w. Construction Aeronautique Tecnam SB339-C5. Prior to installation...</td>
<td>24/08/2023</td>
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<td>10082664</td>
<td>SABENA TECHNICS BGC</td>
<td>SA 319 B, SE 3160 SA 315 B, SA 316 B, SA 316 C</td>
<td>STICK SWITCH GUARD INSTALLATION</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: Original certification basis for the original product + FAR 27.1501 (Amptinv), 27.1529 (Amendment), 27.1581T...</td>
<td>CERTIFICATION PLAN 2139-01-A-CP Rev 0 issued 21/07/23; MASTER DATA LIST 2139-01-A-MLD Rev 0 issued 21/07/23; ROTOCRAFT FLIGHT MANUAL SUPPLEMENT N° 2139-01-A-FLM-F/F Rev 0 issued 21/07/23; SUPPLÉMENT NT AU MANUEL DE VOL N° 2139-01-A-RFMS-F/F Rev 0 issued 21/07/23.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other pre-existing installations change and/or repair will introduce no adverse effect upon the...</td>
<td>24/08/2023</td>
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<td>10082685</td>
<td>PAL AEROSPACE LTD.</td>
<td>REV. 1 TCCL A-LSA22-119D</td>
<td>DHC-8-106 DHC-8-102</td>
<td>Cabin Observer Window Installation/Revision 1. Addition of model DHC-8-106 with SN 275, 314 and retrofit configuration 002-2004121-003-004 for SN 38</td>
<td>Installation shall be performed in accordance with DECA Aviation Engineering Limited Master Data List MDL-20061-58, Revision 2 and it shall be performed in accordance with DECA Aviation Engineering Limited Instructions for Continued Airworthiness/CA20061-58. Revision 2.</td>
<td>Limited to Serial Number 028, 038, 039, 040 and 20061 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other pre-existing installation change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2024</td>
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<td>10082666</td>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td>FAA STC ST04276AT</td>
<td>757-300 757-200</td>
<td>Installation of an Iridium Satellite Communications System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: CS 002 amtld 0: 25, 29, 25.302, 25.561, 25.601, 25.603, 25.771, 25.773, 25.1431, 25.1529, 25.1541C/CS 25.601, 25.305, 25.30...</td>
<td>Master Drawing List 215125-MDL, Revision G, dated 1 August 2023; (A, B, C, D) &gt;Airplane Flight Manual Supplement for the Rockwell Collins IRT-2110/2110Iridium Communication System, 215125-MFL; revision R, dated 30 September 2016; (A, B, C, D) &gt;Instructions for Continued Airworthiness (ICA). Avionics Support Group Inc...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other pre-existing installations change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/09/2023</td>
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<td>Model(s)</td>
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<td>Certification basis</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10082678</td>
<td>AVIONTEC AG</td>
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<td>340A</td>
<td>Installation of a TAS Top Antenna Garmin GA 58. Exchange of the existing WX Radar System by a Garmin GMRX 75 WX Radar Antenna/Receiver Unit.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: §§ 23.887, 23.1306 and/or 23.1398 of CS 23.63d. 4 The requirements for environmental protection and the associated certification basis/ approval were unchanged and remain applicable to this certificate/approval.</td>
<td>Certification Program Doc. No. 14304, dated 22.08.2023; Master Document List: Doc. No. 14304-017, dated 23.08.2023; Aircraft Flight Manual Supplement: Doc. No. 14304-017, dated 19.05.2023; or later revisions of the above listed document(s) approved/applicable to the EASA system.</td>
<td>Certification: This change is limited to Serial Number 340A-0521. This change requires the installation of a Garmin Integrated Electronic Flight Display System G550 TXi installed according to EASA STC 10064764-G; Garmin TAS System GTS 800 installed according to EASA STC 10064737; Prior to installation of this change/re...</td>
<td>29/07/2023</td>
<td>Active</td>
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<tr>
<td>10082682</td>
<td>PATRIA AVIATION OY</td>
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<td>412EP</td>
<td>NVIS Approval</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certification bases/approval were unchanged and remain applicable to this certificate/approval.</td>
<td>- Drawing List: D-412-08-C5-04 revision L, dated 13 July 2023 - Installation document D-412-08-EO-22 revision L, dated 13 July 2023 - Instructions for Continued Airworthiness D-412-08-ICA-03 Revision dated 2 June 2023 - Rotorcraft Flight Manual Supplement D-412-08-RFM-11 Issue 1 Revision dated 2 June.</td>
<td>- The Installation of this modification by third persons is subject to written permission of the approval holder and disposal of the applicable documentation. - Category A operational and emission levels of the product remains applicable to this certificate/approval.</td>
<td>29/07/2023</td>
<td>Active</td>
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<td>10082683</td>
<td>PATRIA AVIATION OY</td>
<td></td>
<td></td>
<td>412, 412EP</td>
<td>Medical Kit Installation / Provisions for the installation of the oxygen bottle, life raft and stretcher.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certification bases/approval were unchanged and remain applicable to this certificate/approval.</td>
<td>- Drawing List: D-412-08-C5-04 revision L, dated 13 July 2023 - Installation document D-412-08-EO-22 revision L, dated 13 July 2023 - Instructions for Continued Airworthiness D-412-08-ICA-02 Issue 2 dated 12 April 2023 - Appendix 4 - Rotorcraft Flight Manual Supplement D-412-08-RFM-12 Issue 1 Revision dated 2 June.</td>
<td>- The Installation of this modification by third persons is subject to written permission of the approval holder and disposal of the applicable documentation. - The Approval holder and emissions of the product remains applicable to this certificate/approval.</td>
<td>30/08/2023</td>
<td>Active</td>
<td>10082683</td>
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<td>10082690</td>
<td>QINETIQ GmbH</td>
<td>REV. 1</td>
<td></td>
<td>PC-12/47E</td>
<td>MX-15 CAMERA INSTALLATION - Installation of a gyro-stabilized camera with increased diameter of up to 16.6 inches to an existing lift system</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certification bases/approval were unchanged and remain applicable to this certificate/approval.</td>
<td>10025 - Standard Configuration. - Installed in accordance with document: Installation Instruction Number 10025-IN-0001, Revision G, dated 6th September 2025 - Operational in accordance with document: Aircraft Flight Manual Supplement, 10025- AFM-0001, Revision E, dated 30th June 2025 - Maintained in accordance with...</td>
<td>- STC Prerequisite: PC-12/47E aircraft equipped with hydraulic landing gear, four bladed Hartzell Propeller HC-E4A/3D/ E1047759K and installed SPECTRE LIFT PLATFORM in accordance with EASA STC 10007720 - (Validated FAA STC SA00634DE-2) Installation of 10038 configuration is limited to MSN 1263 only.</td>
<td>03/11/2023</td>
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<td>10082691</td>
<td>BOMBARDIER Inc.</td>
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<td>TCCA SA23-58</td>
<td>BD-700-1A10 Installation SATCOM System This STC will consist in the installation of a SATCOM system to provide CPDLC and cockpit voice channels for airplanes in a green configuration.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certification bases/approval were unchanged and remain applicable to this certificate/approval.</td>
<td>- Modification Data Summary RAZ-EM7610-29G Revision NC, dated 28 June 2023 - Aircraft Flight Manual Supplement G-FM2300401 Revision Basic, dated 28 June 2023 - Instructions for Continued Airworthiness BAS070580-06-KCA Revision Basic (Issuer) or later revisions of the above listed document(s) approved/applicable to the EASA system.</td>
<td>None</td>
<td>30/08/2023</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10082695</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SA023045E</td>
<td>G48-TG 320 G48</td>
<td>Installation of vortex generators on the wings and tail surfaces</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Drawing Package MA2253, Revision IR, dated 14 February 2014+</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/08/2023</td>
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<td>10082696</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST04035AT-D</td>
<td>GULFSTREAM 200/ GALAXY</td>
<td>Installation of Micro Quick Access Recorder (qQAR)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Gulfstream Aerospace Corporation Index List CEB14208501 Revision*- dated 7 January 2016+</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>31/08/2023</td>
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<td>10082703</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST02091AT-D</td>
<td>767-300</td>
<td>Install a Collins CMU-900 Communications System</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: the following paragraph(s) at a late amendment/C303 until: 12: 25.301, 25.303, 25.305, 25.307, 25.601, 25.603, 25.605, 25.609, 25.1301(a)(x)(ii) - 25.1307(ii) - 25.1309(xii)</td>
<td>TIMCO Master Data List H970980L011 Rev. H, dated 14 March 2001+</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/06/2023</td>
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<td>10082750</td>
<td>SAAB AB</td>
<td>(EASA.A.S.00480)</td>
<td>REV. 2</td>
<td>C-212</td>
<td>Installation of ACAS II and dual mode-S transpondersRev. 1: Transfer of STC Rev. 2 corrects the STC holder name</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Saab Nyce Aero AB, TDM 04-14, Issue 3, dated 2604-07-28 - AFM-supplement 050804r later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>12/06/2023</td>
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<td>10082751</td>
<td>SAAB AB</td>
<td>(EASA.A.S.00694)</td>
<td>REV. 3</td>
<td>C-212</td>
<td>Installation of ACAS II and a single SSR Mode-S transponderRev. 1: Transfer of STC Rev. 2 corrects the STC holder name Rev. 3: Correction of Associated Technical Documentation</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Saab Nyce Aero AB, TDM 04-14, Issue 3, dated 2005-04-11AFM Supplement, AFMS 16, Rev. 4, 11-200504r later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>09/04/2024</td>
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### List of EASA Supplemental Type Certificates

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<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<td>AERODISA S.L.</td>
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<td>AT-802A AT-802</td>
<td>Installation of sensors for development</td>
<td>The Type and OSD Certification Bases for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval within...</td>
<td>AE02002-17M_AFL Rev. A2AED0202-17M_AFMS Rev. A2AED0202-17M_ICA Rev. A10 or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/08/2023</td>
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<td>10082756</td>
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<td>ARCHIEON HOLDINGS, L.L.C.</td>
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<td>FAA STC ST04189AT</td>
<td>767-400ER 767-300F 767-200</td>
<td>Installation of identification system.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/ approval.</td>
<td>Master Data List 130720-MDL Revision B, dated 10 December 2014( ); &gt;Acipena Flight Manual Supplement for Archvii Holdings LLC, 130720-AMFS Revision (R), dated 23 December 2014( ); &gt;instructions for Continued Airworthiness 130720-ICA, Revision B, dated 9 December 2014( ); or later revisions of the EASA AIRCRAFT SYSTEM.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/09/2023</td>
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<td>10082765</td>
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<td>DUNCAN AVIATION, Inc.</td>
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<td>FAA STC ST01974WH</td>
<td>MYSTERIE FALCON 900 S9F960 FALCON 900EX (F900EX)</td>
<td>Installation of Installation of Aviation Clean Air Ionization System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/ approval.</td>
<td>Duncan Aviation, Inc. STC Master Document List, Doc. No. CERTDOC-012909, Rev. A, dated 29 September 2021( ); &gt;Duncan Aviation, Inc. Instructions for Continued Airworthiness, Doc. No. CERTDOC-013216, Rev. A, dated 21 September 2021( ); or later revisions of the above listed document(s) approved/ accepted under the EASA AIRCRAFT SYSTEM.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/09/2023</td>
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<td>DUNCAN AVIATION, Inc.</td>
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<td>FAA STC ST01973WH</td>
<td>FALCON 7X</td>
<td>Installation of Installation of Aviation Clean Air Ionization System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product remain unchanged and applicable to this certificate/ approval.</td>
<td>Duncan Aviation STC Master Document List, Doc. No. CERTDOC-011636, Rev. A dated 7 September 2021( ); &gt;Duncan Aviation, Inc. Instructions for Continued Airworthiness, Doc. No. CERTDOC-013274, Rev. A, dated 7 September 2021( ); or later revisions of the above listed document(s) approved/ accepted under the EASA AIRCRAFT SYSTEM.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/09/2023</td>
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<td>10082767</td>
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<td>NSE</td>
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<td>S2R-T34</td>
<td>DTS V2 for Thruax S2R</td>
<td>The DTSV2 is a tracking system transmitting GPS information through the Iridium network. The installation kit contains all the elements associated to the equipment. DTSV2 such as antenna, coaxial cable, harnesses, circuitbreaker, support etc.</td>
<td>The certification basis for the intended modification is the CS 23 amendment 00 dated November 14, 2003. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any...</td>
<td>SDL-00055181AA Version A11 STC Data List documents AFMS-00055181 AA Version A01, Aircraft Flight Manual Supplement OR later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/09/2023</td>
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<td>SKYTEC AEROSPACE GmbH</td>
<td>18C / 150C Passenger Seating Layout The modification introduces a cabin layout using 50 economy class aisle seats and business class double seats generating a 168 passenger seating configuration. The Type and OSD Certification Bases for the original product are amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: For Boeing 737-800/900ER: CS 25.1, CS 25.23, CS 25.26, CS 25.27 (b), CS 25.29, CS 25.30, CS 25.31, CS 25.32, CS 25.33, CS 25.36, CS 25.37, CS 25.38, CS 25.49. The aircraft shall be operated in accordance with revision A of Skytec Aircraft Flight Manual Supplement AFMS-1.0738C111-002. Maintenance shall be accomplished in accordance with revision A of Skytec Instruction...</td>
<td>The change is defined with revision A of Skytec Master Document List Document MLD-8738C111-002. The aircraft shall be operated in accordance with revision A of Skytec Aircraft Flight Manual Supplement AFMS-1.0738C111-002. Maintenance shall be accomplished in accordance with revision A of Skytec Instruction...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>11/06/2023</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>G2 GALLEY REPLACEMENT Replacement of Galley G2 slim into galley G2 large. The replacement of this galley does not introduce new functionality and new galley is compiled for storage and oven. The Type and OSD Certification Bases for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval within...</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.23, CS 25.30.1, CS 25.30.5, CS 25.30.7, CS 25.32, CS 25.561, CS 25.601, CS 25.603, CS 25.605, CS 25.607, CS 25.609, CS 25.611, CS 25.612</td>
<td>AE22083MDL Issue 1, Master Drawing ListAE22083CA Issue 0, Instruction for Continued Airworthiness; later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The cabin configuration before modification is defined per modification JC/ODS/5854 approved per JC/ODS/5854. Engineering Services Ltd.</td>
<td>08/03/2024</td>
<td>Active</td>
<td>10082773</td>
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<tr>
<td>ALAMO ENGINEERING GmbH</td>
<td>FALCON 2000 Interior Refurbishment with LOPA change/repair refurbishment with seat modification - installation of cabin LED lights and Air Show 500 - Replacement of Al Combi with Alto Cadence coding Speaker - Galley Espresso machine removal with an installation of AC Outlet Speaker - Galley Espresso...</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval within...</td>
<td>MDL22A32AE-01 Master Document List Rev. BCA220323AE-01 Instructions for Continued Airworthiness; Rev. A later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>None</td>
<td>13/06/2023</td>
<td>Active</td>
<td>10082790</td>
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| QUIET TECHNOLOGY AEROSPACE, Inc.  | FAA STC ST04600A Replacement of the existing Honeywell HTF7000 engine upper and lower thrust reverser door assemblies with Quiet Technology Aerospace modified

... | Quiet Technology Aerospace document "HTF 7000 Thrust Reverser Upgrade Master Drawing List," Report No. QTA 7001, Rev. A, dated 26 October 2022, and "HTF 7000 Thrust Reverser Upgrade Master Drawing List," Report No. QTA 7007, Rev. A, dated 20 February 2023. | This modification has been determined to be a no acoustic change (inconformance with 14 CFR §21.93) Prior to installation of the change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse... | 13/06/2023     | Active       | 10082791    |
<p>| STC HOLDING COMPANY, INC.        | FAA STC SA1841N1M Installation of larger windows and the lowering of the seats | The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval. Consulting Aerospace Engineers, Inc. MDL Number GRC-1.1007.&lt;p&gt;&lt;1&gt;Revision H, dated 14 February 1985, &lt;2&gt; prior later revisions of the above listed document(s) approved/accepted under the EASA in accordance with the Technical Implementation Procedures of EU/USA Bilateral Agreement. | Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 18/06/2023     | Active       | 10082808    |</p>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<td>STC HOLDING COMPANY, INC.</td>
<td>A319-171N</td>
<td>1395</td>
<td>FMS 100</td>
<td>412EP</td>
<td>412</td>
<td>Installation of larger windows and the lowering of the seats</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Consulting Aerospace Engineers, Inc. MDL No. Number GRC-1-1007(&lt;v&gt;) &gt;Revision K, 22 August 1988(&lt;v&gt;) &gt; Prior to installation of this change/repair it must be determined that the incompatibility between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/06/2023</td>
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<td>UUGS AERO CONCEPT</td>
<td>A319-171N</td>
<td>1395</td>
<td>FMS 200</td>
<td>412EP</td>
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<td>Replacement of the existing CVR with ASI INNOVATION FMS and EADI_EHSI Installation on UUDS AERO</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): D-01 Medical Evacuation Configuration/50 following paragraph(s) at a later amendment; see below//CS 25.1; 25.29-35-303; 25-305; 25-305; 2...</td>
<td>Master Data List UAC22M002-MDL01 Rev 0 dtd. 06 Sep 2023/Aircraft Flight Manual Supplement UAC22M002-JFA5901 Rev. 2 dtd. 28Jul2023/Instructions for Continuous Airworthiness UAC22M002-ICA01 Rev. 0 dtd. 20Mar2023 or later revisions of the above listed document(s) approved/ accepted under the EASA system.</td>
<td>- Prior to installation of this change/repair it must be determined that the incompatibility between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/09/2023</td>
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<td>ALAMO ENGINEERING GmbH</td>
<td>AS 355 N</td>
<td>412EP</td>
<td>FMS 100</td>
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<td>Laser and antenna activation on AS355N equipped with modification ASIFAM-1782 No. 02230001.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): Special Condition Doc. No. 5C-PS1 1395-01 the following paragraph(s) at a later amendment; CS 25.23, CS 25-301, CS-325.303, CS 25 305, CS 2...</td>
<td>Master Data List Doc. No. MDL22A43AE-01 Revision A, dated 13 September 2023/Aircraft Flight Manual Supplement FMS22A43AE-01 Revision A, dated 15March 2023 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the incompatibility between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/09/2023</td>
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<td>ASI INNOVATION</td>
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<td>FMS 100</td>
<td>412EP</td>
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<td>FMS and EADI_EHSI Installation on B412</td>
<td>The Type and OSD Certification Bases (CS) for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): Special Condition Doc. No. 5C-PS1 1395-01 the following paragraph(s) at a later amendment; CS 25.23, CS 25-301, CS-325.303, CS 325...</td>
<td>- Prior to installation of this change/repair it must be determined that the incompatibility between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>- Prior to installation of this change/repair it must be determined that the incompatibility between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>PATRIA AVIATION OY</td>
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<td>FMS 100</td>
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<td>FMS and EADI_EHSI Installation on B412</td>
<td>The Type and OSD Certification Bases (CS) for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition (s): Special Condition Doc. No. 5C-PS1 1395-01 the following paragraph(s) at a later amendment; CS 25.23, CS 25-301, CS-325.303, CS 325...</td>
<td>- Prior to installation of this change/repair it must be determined that the incompatibility between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>- Prior to installation of this change/repair it must be determined that the incompatibility between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>Date of Issue</td>
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<td>JET AVIATION AG</td>
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<td>10082918</td>
<td>Replacement of installed CMA-2102 SBB antenna with HGA-7001 antenna.</td>
<td>The Type and OSD Certification Bases for the original product are amended by the following:</td>
<td>Master Data List JBL-M-03868MDL-001 Issue 8: Instructions for Continued Airworthiness JBL-M-03868ICA-010 Issue 8 or later revisions of the above listed document(s) approved/revised/under the EASA system.</td>
<td>This certificate covers only the structural provisions for this antenna installation, internal lining and installation of any electronic hardware need to be covered by a separate airworthiness approval. Prior to installation of this change/repair it must be determined that the interrelationship between...</td>
<td>21/08/2023</td>
<td>Active</td>
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<td>PATRIA AVIATION OY</td>
<td>AB412 EP</td>
<td>AB412</td>
<td>412EP</td>
<td>412</td>
<td>Rotorcraft Modular Equipment Rack – RMER / Installation of the modularityisation device, which allows flexible fixations of multiple different loads and cargo in cabin.</td>
<td>The Type and OSD Certification Bases for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval within...</td>
<td>-Drawing List D-412-15- CS-01 revision C dated 1 September 2023. Installation document D-412-15-EO-01 revision C dated 1 September 2023: Instructions for Continued Airworthiness D-412-15-ICA-01 Revision 1 dated 1 September 2023. Rotorcraft Flight Manual Supplement D-412-15-RFA-01 Issue 05 date...</td>
<td>The Installation of this modification by third persons is subject to the approval holder and the approval of the approved appropriate documentation. Patria minor change D-412-15-DA-02 Provision for Rotorcraft Modular Equipment Rack is accomplished. Prior to installation of this change/app...</td>
<td>21/08/2023</td>
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<td>AERODISA S.L.</td>
<td>A350-941</td>
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<td>10082852</td>
<td>New Waste Compartment Door on A350-900 aircraft Installation of a cabinet's door on lavatories in A350-900 aircraft</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval within...</td>
<td>Airworthiness Compliance Program – AED0202-16M_PCA - Installation of cabinet’s door on A350 aircraft/rev A0 dated 12 September 2023: MASTER Data List - AED0202-16M_MDL - Installation of a cabinet’s door on A350 aircraft/rev A0 dated 21 September 2023: later revisions of the above listed document(s).</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/06/2023</td>
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<td>AEROSPACE ENGINEERING GmbH</td>
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<td>10082855</td>
<td>FMS Upgrade UNS-1L to UNS-1Lw including RNP1</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval.</td>
<td>Instructions for Continued Airworthiness JBL-M-03894MDL-001 Issue 8: Revisions of the above listed document(s) approved/revised/under the EASA system.</td>
<td>Limited to aircraft S/N 265: Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/06/2023</td>
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<td>GULFSTREAM CORPORATION</td>
<td>FAA STC</td>
<td>ST03220AT-D</td>
<td>GY-SP</td>
<td>10082918</td>
<td>Installation of Pocket Door Lockout Mechanism</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged remain applicable to this certificate/approval.</td>
<td>Gulfstream Aerospace Corporation Index Card GS1-5188000; Revision C dated 2 May 2006: ( \Rightarrow ) GA Airplane Flight Manual Supplement, GS1-5188000 Issue 5. Revision NC, dated 4 May 2006: ( \Rightarrow ) For later revisions of the above listed document(s) approved/revised/approved/revised under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/06/2023</td>
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03/07/2024
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<td>10082926</td>
<td>AERODATA AG</td>
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<td>EMB-550</td>
<td>Installation of Aerodata AG Flight Inspection System AD-AFIS-1350.</td>
<td>The Type and OSD Certification Bases (CB) for the original production-certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Master Document List Installation of AD- AFIS-1350, document/MLD- A/C-948Rev 9, dated 19 September 2023</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/03/2023</td>
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<td>10082933</td>
<td>SST FLUGTECHNIK GmbH</td>
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<td>KODIAK 100</td>
<td>Single Hatch Installation</td>
<td>The Type and OSD Certification Bases (CB) for the original production-certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Installed in accordance with EO-KOD100-9506-01 dated 30/8/2023; maintained in accordance with MMS-KOD100-9506-01 dated 19/8/2023; operated in accordance with AFMS-KOD100-9506-01 dated 22/9/2023.</td>
<td>Applicable to Aircraft Serial Numbers: up to and incl. 100-07274 Not compatible with the optional External Cargo Compartment (ECC) Not compatible with the optional TKD Ice Protection System.</td>
<td>02/10/2023</td>
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<td>10082937</td>
<td>HONEYWELL INTERNATIONAL Inc</td>
<td>FAA SAH19SNY</td>
<td>SEE APPROVED MODEL LIST (AMT NO. 89900019-110 - 15/07/2020)</td>
<td>See Approved Model List</td>
<td>Installation of the KFC 230 Avionics Automatic Flight Control System (AFCS) upgraded from KFC 200 FCS.</td>
<td>The Type and OSD Certification Bases (CB) for the original production-certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Approved installation of a KFC 230 Flight Control System (AFCS) asspecified by AML No. 89900019-110 is a prerequisite to this STC. The ICA and its contents must be incorporated into the operator's maintenance program.</td>
<td>16/10/2023</td>
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<td>10082968</td>
<td>BOEING IRELAND LIMITED</td>
<td>REV. 1</td>
<td></td>
<td>Boeing 787-9 – QTR 3U Business Class Suites Installation/Introduction of a 30U/24Y passenger seating layout through the installation of 30 Business Class (B/C) seats and associated mini-suite furniture and front row monuments in the forward passenger seating zone between doors 1 and 2, Rev. 1.</td>
<td>The Type and OSD Certification Bases (CB) for the original production-certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>Boeing 787-9 – QTR 3U Business Class Suites Installation/Introduction of a 30U/24Y passenger seating layout through the installation of 30 Business Class (B/C) seats and associated mini-suite furniture and front row monuments in the forward passenger seating zone between doors 1 and 2, Rev. 1.</td>
<td>- As per BIL-00180-AFM01, §1.1. – Implementation of STC 10082966 Rev 1 or later approved revisions replaces Alternate Method of Compliance for EASA AD 2023-0221 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10082976</td>
<td>PMA ENGINEERING</td>
<td>CAB-04740 – 213 Y/C seats installation</td>
<td>737-9</td>
<td>The Type and OSD Certification Bases (CB) for the original production-certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>The Type and OSD Certification Bases (CB) for the original production-certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
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<td>SKYTEC AEROSPACE GmbH</td>
<td>737-8</td>
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<td>Iridium Antenna Installation</td>
<td>The Type and OSD Certification Bases (CB) for the original product are applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>The change is defined with revision A of Skytec Master Document List/Document M-0362MDL-00Maintenance CE80-900, Issue LT0077-MDL0-0100, dated 11 August 2022. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/10/2023</td>
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<td>10082982</td>
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<td>737-8</td>
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<td>Iridium Antenna Installation</td>
<td>The Type and OSD Certification Bases (CB) for the original product are applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>The change is defined with revision B of Iridium Antenna Installation Master Data List/Document M-0362MDL-00Maintenance CE80-900, Issue LT0077-MDL0-0100, dated 11 August 2022.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>AVMAX AVIATION SERVICES Inc.</td>
<td>TCCA SA13-49</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Cockpit Stowage Box Installation</td>
<td>The Type and OSD Certification Bases (CB) for the original product are applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>The change is defined with revision B of AVMAX Aviation Services Inc. Master Data List/Document M-0362MDL-00Maintenance CE80-900, Issue LT0077-MDL0-0100, dated 11 August 2022. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>10082995</td>
<td>NEXTANT AEROSPACE, L.L.C.</td>
<td>FAA STC ST8360CH</td>
<td>GVI (G650)</td>
<td>Installation of a SpaceX Starlink Aviation System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>The change is defined with revision A of Nextant Aerospace Master Data List/Document M-0362MDL-00Maintenance CE80-900, Issue LT0077-MDL0-0100, dated 11 August 2022. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/10/2023</td>
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<td>10083025</td>
<td>PRATT &amp; WHITNEY ENGINE SERVICES, IN</td>
<td>FAA STC SA10154DE</td>
<td>1900D/1900C</td>
<td>Installation of Pratt &amp; Whitney Engine Services FAST System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>The change is defined with revision A of Pratt &amp; Whitney Engine Services FAST System Master Data List/Document M-0362MDL-00Maintenance CE80-900, Issue LT0077-MDL0-0100, dated 11 August 2022. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/10/2023</td>
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<td>10083025</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>10083029</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>737-300, 737-400, 737-500</td>
<td>STC 320</td>
<td>Installation of Skytrac Iridium Certus on pressurized and non-pressurized compartments. Integration of an additional or alternative interconnectivity requirement.</td>
<td>The Type and OSD Certification Bases (CB) for the original production systems applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
<td>Compliance to STC 3000115-COR Issue 2, Engineering Order 3000115-EO Issue 1. Maintenance Manual Supplement: 3000115-MM Issue 1.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/10/2023</td>
<td>Active</td>
<td>10083029</td>
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<tr>
<td>10083031</td>
<td>BOEING IRELAND LIMITED</td>
<td>777-300ER</td>
<td></td>
<td>Saudi Arabian Airlines, 777-300ER Cabin Reconfiguration</td>
<td>The aircraft configuration is changed from a production 30U (Business Class) + 38Y (Economy Class) passenger seat layout to a 12Y (Business Class) + 43Y (Economy Class) passenger seat layout.</td>
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<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/10/2023</td>
<td>Active</td>
<td>10083031</td>
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<tr>
<td>10083051</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>737-800</td>
<td></td>
<td>Boeing 737-800 (186Y/C)</td>
<td>The Type and OSD Certification Bases (CB) for the original production systems applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
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<td>18/10/2023</td>
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<td>10083051</td>
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<td>10083061</td>
<td>HELIKOPTER AIR TRANSPORT GESELLSCHAFT b. H.</td>
<td></td>
<td></td>
<td>Boeing 737-800 (186Y/C)</td>
<td>Boeing 737-800 (186Y/C), into a single-class cabin configuration (186 Y/C).</td>
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<tr>
<td>10083064</td>
<td>AERODATA AG</td>
<td>B300</td>
<td></td>
<td>Installation of AD-APIS-2100 Main Item: - Integration of an electrical load switch for FIS power supply - Integration of tether release on pressurized and non-pressurized compartments - Integration of one operator and one equipment container in the cabin. - Decrease of passenger seats - Integration of additional...</td>
<td>The Type and OSD Certification Bases (CB) for the original production systems applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
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<td>19/10/2023</td>
<td>Active</td>
<td>10083064</td>
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<tr>
<td>10083072</td>
<td>ALAMO ENGINEERING GmbH</td>
<td>DORNIER 228-212</td>
<td></td>
<td>Installation of Skytrac Iridium Certus Satcom</td>
<td>The Type and OSD Certification Bases (CB) for the original production systems applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/approval.</td>
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<td>20/10/2023</td>
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<td>EASA STC no.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10083134</td>
<td>SABENA TECHNICS BOD</td>
<td>737-800</td>
<td></td>
<td>Relocation of a Low Frequency Underwater Locating Device (LF-ULD)</td>
<td>The Type and OSD Certification Bases (CB) for the original produc stomets remain applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the product are unchanged remain applicable to this certificate/approval w...</td>
<td>Master Document List MDL-5-00-419 Issue Approval for Continued Airworthiness ICA- S0036-00-002 Issue for later revisions of the above listed document(s) approved/ accepted/ed by the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherpreviously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/10/2023</td>
<td>Active</td>
<td>10083134</td>
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<tr>
<td>10083139</td>
<td>TELEDYNE UK LIMITED</td>
<td>A300</td>
<td></td>
<td></td>
<td>Installation of Wireless GroundLink Upgrade of existing Wireless Quick Access Recorder (WQAR) unit to support continued network operation legacy mobile phone networks are shut down.</td>
<td>The Type and OSD Certification Bases for the original product as amended the following additional or alternative airworthiness requirements the following paragraph(s) at a later amendment: CS25.899 Amdt 2 The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the product are unchanged remain applicable to this certificate/approval.</td>
<td>-Teledyne Instructions for Continued Airworthiness, Doc Ref MOD-ICA-043, Rev 01, Dated 19 January 2023-Teledyne Master Drawing List, Doc Ref MOD-MDL-043, Rev 01, Dated 23 June 2023or later revisions of the above listed document(s) accepted/ed by the EASA system.</td>
<td>EASA STC 10033914 (which is a validation of FAA STC ST01778LA installation of Wireless GroundLink) is a pre-requisite before reinstating this STCPrior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherpreviously installed c...</td>
<td>27/10/2023</td>
<td>Active</td>
<td>10083139</td>
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<tr>
<td>10083196</td>
<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC SA01304WI</td>
<td>SR22T</td>
<td>Installation of the Concorde CB24-11M or CB24-11MXC Lead Acid Battery (&gt; or &gt; or RG24-11M or RG24-15M VRiSLA Battery.</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Concorde Master Drawing List, Dwg. No. 5-0044, dated 17 June 2004, or Dwg. No. 5-0044 Rev A, dated 7 September 2011&gt; or (&gt;or Concorde Instructions for Continued Airworthiness, Dwg. No. 5-0044, Revision F, dated 11 March 2011&gt; or (&gt;for the RG series batteries&lt; or for later revisions of the above labels&lt;.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherpreviously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/11/2023</td>
<td>Active</td>
<td>10083196</td>
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<tr>
<td>10083197</td>
<td>BOEING IRELAND LIMITED</td>
<td>737-8</td>
<td>737-8 Class 2 Crew Rest Seat Installation of new design single cabin attendant seat capable of meeting Class 2 crew rest specifications.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>Change / Repair Approval: 737-8 Class 2 Crew Rest Seat, BIL-0006, APP.Revision 00, dated 1st November 2023 Master Minimum Equipment List Supplement, &quot;NEOS 737-8 Class 2 Crew Rest Seat&quot;, BIL-0026-MMEL01 R01, dated 8th September 2023 Flight Attendant Manual (FAM) Supplement, &quot;NEOS 737-8 Class 2 Crew...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherpreviously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/11/2023</td>
<td>Active</td>
<td>10083197</td>
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<td>10083201</td>
<td>J &amp; C AEROSPACE DESIGN ANDCERTIFICATI ON SERVICES EUS LTD</td>
<td>REV. 1</td>
<td>737-800</td>
<td>Cabin reconfiguration with new Safran Model Z10D Series economy class passenger seats equipped with ISPS.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the original product remain unchanged and remain applicable to this certificate/approval.</td>
<td>MDL referenced JCAERO_MDL_0879_1SS 1.1 dated May 24th 2024- AM5 Supplement referenced JCAERO_AMM_0879_1SS 1.0 dated June 1st 2023or later revisions of the above listed document(s) approved/ed by the EASA system.</td>
<td>-Change only applicable to MSHs listed in the Certification/here reference JCAERO/AM5/0879Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any otherpreviously installed change and/ or repair will introduce no adverse effect upon...</td>
<td>06/04/2024</td>
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<td>10083201</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10083204</td>
<td>DUNGAN AVIATION, Inc.</td>
<td>FAA STC SA1176W1</td>
<td>1900D</td>
<td>1900C</td>
<td>Installation of Collins TCAS-4000 Traffic Alert and Collision Avoidance System (TCAS-II)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>(1) Duncan Aviation Document List 10599100+ (16) Revision A, dated 5 June 2003</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/11/2023</td>
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<td>10083204</td>
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<td>10083205</td>
<td>STERLING CORPORATION/ B&amp;J STERLING HELICOPTER</td>
<td>FAA STC ST04179NY</td>
<td>GVI (G650)</td>
<td>GVI-X, GVI-SP</td>
<td>Installation of True Blue Power TFC4000 AC to AC Converter</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval.</td>
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<td>10083206</td>
<td>ELLIOTT AVIATION TECHNICAL PRODUCTS DEVELOPMENT, LLC</td>
<td>FAA STC ST02217CH</td>
<td>650</td>
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<td>Installation of a Honeywell CAS 67A Traffic Alert and Collision Avoidance System (TCAS II)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Elliott Aviation Master Data List, Report No. MDS-2515-000, Revision A, dated 23 November 2005</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/11/2023</td>
<td>Active</td>
<td>10083206</td>
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<td>10083207</td>
<td>S4A SOLUTIONS FOR AVIATION, S.L</td>
<td>CL-600-2D4 (RJ SERIES 900)</td>
<td>Low Frequency ULB Installation on CRJ600</td>
<td></td>
<td></td>
<td>The Type and OSD Certification Bases for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certification noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>604_166-16-02 Master Data List 26-10-2023604_166-12-30 Maintenance Manual Supplement (ICA Edition) 2nd later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/11/2023</td>
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<td>10083208</td>
<td>AIRBUS S.A.S.</td>
<td>A320-271NX</td>
<td>Airbus STC Solutions MOD A0257 - INSTALL CUSTOMIZED MCD</td>
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<td>STC Holders</td>
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<td>Limitations/Conditions</td>
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<tr>
<td>ELIOTT AVIATION TECHNICAL PRODUCTS/DEVE LPMENT, LLC</td>
<td>FAA STC ST02218CH</td>
<td>Installation of a Universal Avionics EFI-890R Electronic Flight Instrument System (EFIS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>Elliott Aviation Master Data List, Revision A, dated 23 November 2005 (v.1)</td>
<td>Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/11/2023</td>
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<td>SKY SERVICE F.B.O. INC.</td>
<td>TCCA SA22-30 ISSUE 1 (VARIANT)</td>
<td>Challenger Ku Band Satcom System: This modification covers a Ku Band Satcom Installation. This system is a standalone providing Wi-Fi in the cabin.</td>
<td>The installation shall be performed in accordance with Antuvision Design Services Ltd Master Drawing List MDL21033, Revision N and/or C. The aircraft shall be operated in accordance with Skyservice F.B.O. Flight Manual Supplement FMS21033-1.</td>
<td>EASA STC 10066792 shall be installed as a prerequisite. Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/11/2023</td>
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<tr>
<td>ELIOTT AVIATION TECHNICAL PRODUCTS/DEVE LPMENT, LLC</td>
<td>FAA STC ST02216CH</td>
<td>Installation of a Universal Avionics Terrain Awareness and Warning System (TAWS)</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>Elliott Aviation Master Data List, Revision A, dated 23 November 2005 (v.1)</td>
<td>Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/11/2023</td>
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<td>3S CERTIFICATION, LLC</td>
<td>FAA STC SA0017W1</td>
<td>Installation of the Parker-Meggitt iPRESS™ System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unaltered and remain applicable to this certificate/ approval.</td>
<td>3S Certification, Master Data List (MDL) 3518-C36-1001, Revision IR, dated 1 August 2023 (v.7)</td>
<td>Approval of this change in type design applies only to the aircraft listed on Approved Model List 3518-C36-1001. Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>09/11/2023</td>
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<td>10083255</td>
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<td>GDC ENGINEERING GmbH</td>
<td>A319-111</td>
<td>44 B/C – 6 YIC Cabin Reconfiguration: This modification introduces 22 Business Class Double Seats and 2 Economy Class Triple Seats resulting in a 50 Passenger Seating Configuration Layout.</td>
<td>The modification is defined in accordance with GDC Engineering Master Document Register A319K-MDR-25000, Issue B. The aircraft shall be maintained in accordance with GDC Engineering Instructions for Continuation Airworthiness &amp; Operator Manual Supplement Document A319K-ICA-BYD, Revision 0. The OSD-Cabin Crew...</td>
<td>Prior to installation of this change/repair, it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/11/2023</td>
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<td>Foreign/NAA STC (if applicable)</td>
<td>Model</td>
<td>Description</td>
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<td>Certification date</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>Garmin Inc</td>
<td>SA01695SE</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Installation of antenna model identified in the Master Data List</td>
<td>The Certification Basis for the original product remains applicable unless certificate was revoked, except where amended by additional or later amendments if included on FAA STC SA01695SE. The requirements for environmental protection and the associated Identification Noise and/or emissions levels of the original product.</td>
<td>Garmin Supplemental Type Certificate Master Data List, FAA 008-00373-00 Revision D, dated 21 August 2000</td>
<td>approved/accepted on behalf of EASA in accordance with the Technical Implementing Procedures of EU-US Bilateral Agreement.</td>
<td>This STC is for physical installation only. For operational Airworthiness approval refer to the FAA STC Master Data List.</td>
<td>14/11/2023</td>
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<td>10083026</td>
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<td>Sabena Technics BOC</td>
<td>A320-251N</td>
<td>PORTABLE ELT INSTALLATION</td>
<td>Model Data List 41151-01-0 MLD-Rev00/Instructons for Continued Airworthiness 41151-01-1-ICA Rev00/later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements.</td>
<td>Master Data List 41151-01-A-MLD Rev00/Instructions for Continued Airworthiness 41151-01-1-ICA Rev00/later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/11/2023</td>
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<td>L3 Harris Aviation Products Inc</td>
<td>FAA ST0721LA</td>
<td>CL-600-2024 (RJ SERIES 000) CL-600-2B19 (RJ SERIES 100)</td>
<td>Installation of SRV/VR25 Cockpit Voice Recorder (CVR) with Integrated Recorder Independent Power Supply (IRPS)</td>
<td>Master Data List 8900190-00 Rev. 30</td>
<td>Master Data List 8900190-00 Rev. 30</td>
<td>Prior to installation of this change/repair it must be determined the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/11/2023</td>
<td>Active</td>
<td>10083028</td>
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<td>Alamo Engineering GmbH</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>ADS-B Out Collins TDR-94/4 IWAAS GPS/ADS-B.0 Out upgrade in compliance with the latest RTCA DO-268B with Rockwell Collins Proline 4 Avionic suite.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements.</td>
<td>Master Data List M23A37AE-01, Revision A. The aircraft shall be maintained in accordance with Alamo Engineering Instructions for Continued Airworthiness document ID 23A37AE-01.</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/11/2023</td>
<td>Active</td>
<td>10083029</td>
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<tr>
<td>Aircraft Development and Systems Engineering (ADSE) B.V</td>
<td>737-800</td>
<td>Certification of the Boeing 737-800 as PED tolerant</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements.</td>
<td>The change is defined with issue 2.0 of ADSE Design Definition List 250-037. The aircraft shall be operated in accordance with issue 0 of ADSEFAIRLINE Flight Manual Supplement documents 23-DO-035/0, later revisions of the above listed document(s) approved/accept under the EASA system.</td>
<td>The change is defined with issue 2.0 of ADSE Design Definition List 250-037. The aircraft shall be operated in accordance with issue 0 of ADSEFAIRLINE Flight Manual Supplement documents 23-DO-035/0, later revisions of the above listed document(s) approved/accept under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/11/2023</td>
<td>Active</td>
<td>10083030</td>
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<tr>
<td>Eichelsdoerfer GmbH Flugzeugbau</td>
<td>SEE FAA APPROVED MODEL LIST</td>
<td>Covering and reconditioning of an E1A-1 simxuepse with plastic cover</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements.</td>
<td>Technische Mitteilung Nr. 010-23 Rev.00 Issue dated 14/06/2023.</td>
<td>Prior to installation of this change/repair it must be determined that the change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/11/2023</td>
<td>Active</td>
<td>10083031</td>
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### List of EASA Supplemental Type Certificates

<table>
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<tr>
<th>STC Holder</th>
<th>STC ref.</th>
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<th>Certification basis</th>
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<th>System Record</th>
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<tbody>
<tr>
<td>Q.C.M. DESIGN GmbH</td>
<td>10083329</td>
<td>LEARJET MODEL 45</td>
<td>Installation of FDR and SSCVR Upgrade</td>
<td>The Type and OSD Certification Bases (CB) for the original product, associated certificate noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>LAX5-CDNM206-1M1-ICA</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other preinstalled change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/11/2023</td>
<td>Active</td>
<td>10083329</td>
</tr>
</tbody>
</table>

| AVMAX AVIATION SERVICES Inc.             | 10083336 | DHC-8-315        | T744 Lithium Battery System Installation/Modification replaces the existing aircraft rechargeable Nickel-Cadmium batteries with the Mid-Continent Instrument T744 rechargeable Lithium-ion battery. | The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: 25.899, 25.1301, 25.1303, 25.1302, 25.1325, 25.1317, 25.1319, 25.1323, 25.1333, 25.1335, 25.1337, 25.1339, 25.1341. | | Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other preinstalled change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 23/11/2023 | Active | 10083336     |

| Q.C.M. DESIGN GmbH                       | 10083348 | S60XL            | Update of ICA for CAT II Operation removing bench check requirement.       | The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: 25.899, 25.1301, 25.1303, 25.1302, 25.1325, 25.1317, 25.1319, 25.1323, 25.1333, 25.1335, 25.1337, 25.1339, 25.1341. | QC3 Compliance Report, CR-C550-QC3M4-001, Revision 1, dated 01 November 2023. | Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other preinstalled change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 24/11/2023 | Active | 10083348     |

| GULFSTREAM AEROSPACE CORPORATION         | 10083381 | FAA STC ST0434AT-D | Installation of Satcom Direct Router (SDR) Databus Wire Harness | The Certification Basis for the original product remains applicable to this certificate/approval. | Gulfstream Aerospace Corporation Index List GDL100000069, Revision A, dated 4 May 2023 | Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other preinstalled change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 30/11/2023 | Active | 10083381     |

| AMAC AEROSPACE SWITZERLAND AG            | 10083382 | A321-271NX-272NX | WLAN Activation This Modification activates the Wireless Local Area Network (WLAN) function of a preinstalled WLAN system/Router interfaced with WAPs in the cabin emitting in 2.4GHzand/or 5GHz Frequency Band and a coexisting WLAN ON/OFF switch. | The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: 25.899, 25.1301, 25.1303, 25.1302, 25.1325, 25.1317, 25.1319, 25.1323, 25.1333, 25.1335, 25.1337, 25.1339, 25.1341. | Configuration 1: The Change is defined with Revision A of Master Data List document MDL- A321-1.4430. The aircraft shall be operated in accordance with Revision A of Aircraft Flight Manual Supplement document AFMS: A321-1.4430. Maintenance shall be done in accordance with Revision A of Instructions for... | Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other preinstalled change and/or repair will introduce no adverse effect upon the airworthiness of the product. | 30/11/2023 | Active | 10083382     |

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**WLAN Activation This Modification activates the Wireless Local Area Network (WLAN) function of a preinstalled WLAN system/Router interfaced with WAPs in the cabin emitting in 2.4GHz and/or 5GHz Frequency Band and a coexisting WLAN ON/OFF switch.**

**Configuration 1:**

- The Change is defined with Revision A of Master Data List document MDL-A321-1.4430. The aircraft shall be operated in accordance with Revision A of Aircraft Flight Manual Supplement document AFMS: A321-1.4430. Maintenance shall be done in accordance with Revision A of Instructions for...
<table>
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<tr>
<th>Date of Issue</th>
<th>STC Holder</th>
<th>STC No.</th>
<th>Description</th>
<th>Certification basis</th>
<th>Limited/Conditions</th>
<th>System Record</th>
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<tr>
<td>30/11/2023</td>
<td>GULLSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST04347AT-D</td>
<td>Installation of Salton Direct Plane Simple Ku System</td>
<td>Gullstream Aerospace Corporation Index List GD110000076 Revision (-&gt;) dated 08 September 2023 &lt;-&gt;</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product/or existing network, hard wired or WiFi, router operational.</td>
<td>-</td>
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<tr>
<td>30/11/2023</td>
<td>MAGNETIC MRO AS</td>
<td>EAU 190-200 STD EAU 190-200 LR EAU 190-200 IGW</td>
<td>Cabin Rec configuration This reconfiguration modifies the cabin by installing a Galley G2, relocating the FWD RH Storage and accomplishing related interconnects.</td>
<td>The change is defined with Issue 2 of Magnetic MRO Master Data List MRO-C-0215-AMM-01 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>30/11/2023</td>
<td>KITCHENER AERO AVIONICS LIMITED/tda MD CANADA MOD CENTER</td>
<td>TCCA O-LSA20-117/D</td>
<td>UASC TAWIS Software Upgrade (from SCN 10.4 to SCN 10.8).</td>
<td>Avionics Design Services Ltd. Certification Plan CP0053-1. Issue 1 dated 13 October 2020</td>
<td>Limited only to S/N 560-5101. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>-</td>
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<tr>
<td>01/12/2023</td>
<td>AMAC AEROSPACE SWITZERLAND AG</td>
<td>A320-232</td>
<td>Cabin Rec configuration with Commercial Operation This reconfiguration modifies the aircraft equipped with EASA STC10050094 allowing private and commercial operation including increasing the maximum passenger seating capacity to 31 for private configuration and limiting the passenger seating capacity too...</td>
<td>The Change is defined with Revision A of Master Data List documentMLD-HP03-2500. The aircraft shall be operated in accordance with Revision C of Airplane Flight Manual Supplement document AFMS-HP03-2500. Maintenance shall be done in accordance with Revision A of Instructions for Continued Airworthiness d...</td>
<td>EASA STC 10050094 Rev 1 is a pre-requisite to this change. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>01/12/2023</td>
<td>AIRBORNE TECHNOLOGIES GmbH</td>
<td>407</td>
<td>Bell 407 - PNP Survey Cabin Configuration This certificate grants approval for the installation of a Tactical Workstation in the standard passenger cabin in lieu of the LH aft facing seat.</td>
<td>Document List, Document No. STC-186-G-DOO Rev 0 Drawings List, Document No. STC-186-G-DGW Rev. 0 dated 28.11.2023 Certification Program, Document No. STC-186-G-CP Rev. 2 dated 31.10.2023 FM Supplement, Document No. STC-186-G-AMF Rev. 0 dated 27.10.2023 Maintenance Manual Supplement, Document No. ST...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAAC STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
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<td>10083402</td>
<td>FOKKER TECHNIEK B.V.</td>
<td>REV. 3</td>
<td></td>
<td>GV-SP</td>
<td>STC_Fokker_AF-M-S update G550-JCG</td>
<td>The Type and OSD Certification Bases (CB) for the original production apparatus applicable to this certification/approval. This certification/approval involves a change with an impact to the noise database. The environmental protection requirements amended as follows: ICAO Annex 16 Volume 1 Edition 8. Amen...</td>
</tr>
<tr>
<td>10083410</td>
<td>SAAB AB</td>
<td>REV. 1</td>
<td></td>
<td>35A</td>
<td>Installation of ACAS II (Collins TCAS 4000) and dual SSR Mode-Stransponders (Collins ZIR 94)</td>
<td>The Type and OSD Certification Bases (CB) for the original production apparatus applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certification/approval.</td>
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<tr>
<td>(EASA.A.S.00255)</td>
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<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>10083426</td>
<td>MT-PROPPELLER ENTWICKLUNG GmbH</td>
<td>REV. 3</td>
<td></td>
<td>162P, 162Q, F162P, 162Q, 162H, 162K, 162L, 162M</td>
<td>Installation of MTV-9-D/290-68( ) propeller on Cessna (F)182 series equipped with Teledyne Continental IO-550-D engine.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative environmental protection requirements - The following paragraph(s) is/are added: CS-36 Am6r 6, FAR 36/Am6r 36-31. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certification/approval.</td>
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<tr>
<td>10083442</td>
<td>STERLING CORPORATION/ B/A STERLING HELICOPTER</td>
<td>REV. 1</td>
<td></td>
<td>AW139</td>
<td>Installation of Ourenex Corporation Air IP system</td>
<td>The Certification Basis for the original product remains applicable to the certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain unchanged.</td>
</tr>
<tr>
<td>10083466</td>
<td>VHI AEROSPACE Inc.</td>
<td>REV. 1</td>
<td></td>
<td>S-92A</td>
<td>Hatch Close Out Panel Installation. The hatch close out panel installation covers the exposed emergency exit assembly mechanism when equipped with a bubble window installation.</td>
<td>The Type and OSD Certification Bases (CB) for the original production apparatus applicable to this certification/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certification/approval.</td>
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<tr>
<td>STC Holder</td>
<td>Associated Technical</td>
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<td>Status</td>
<td>System Record</td>
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<tr>
<td>L3 AVIATION PRODUCTS, INC</td>
<td>Installation of L3Harris Aviation Products SRF/IRVS Model 7100 Cockpit/Voice Recorder (CRV)</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): 25-660-SC, Non-Rechargeable Lithium Battery Installation. The following paragraph(s) at a later amendment: CS 25.981 (a)(3) at AMDT 1.C.</td>
<td>11/12/2023</td>
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<tr>
<td>WEST STAR AVIATION, LLC.</td>
<td>Installation of Honeywell FANS-1/A+ Datalink Communication System</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Special Condition(s): F25-1333-01 Non-rechargeable Lithium Battery Installation, Special Condition(s): F-34 Flight Recorders and Data Link Recording. The following paragraph... West Star Aviation Master Document(s) (..).</td>
<td>18/12/2023</td>
<td>Active</td>
<td>10083487</td>
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<tr>
<td>GDC ENGINEERING GmbH</td>
<td>Installation of L3Harris Aviation Products Index of MasterData List Doc. No. 8860079. Revision: The aircraft shall be operated in accordance with L3Harris Aviation Airplane Flight Manual Supplement Doc. No. 8210963-900. Rev...</td>
<td>Applicable to S/N: 750-0001 through 750-0500. Prior to installation of this change/repair it must be determined that the...</td>
<td>14/12/2023</td>
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<td>10083528</td>
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<tr>
<td>AERODATA AG</td>
<td>Installation of Aerodata AG Flight Inspection System AD-AFIS-1320 Aerodata Modification Ident: CB80A-O19Revision 1 of this Approval corrects the information about OSD FC limitations section.</td>
<td>The performance of Aerodata Flight Inspection System AD-AFIS-1320 has not been assessed by EASA. This STC cannot apply to...</td>
<td>10/04/2024</td>
<td>Active</td>
<td>10083541</td>
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<tr>
<td>DOA AESGLOBAL LIMITED</td>
<td>189Y Passenger Seating LayoutThis modification introduces acatin layout using 63 triple economyclass seats generating a 189 Passenger seating configuration without acommodating...</td>
<td>The change is defined with issue 3 of AESGLOBAL Master Data List AES-737-7282MDL Maintenance shall be accomplished in accordance with issue 1 of AESGLOBAL Instructions for Continuous Airworthiness Document AES-TR-3892The OSD-Cabin Crew Data (CCD) approved per the CS-CCD issue 2, is contained by...</td>
<td>15/12/2023</td>
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<td>STC Holder</td>
<td>STC Number</td>
<td>STC STC No.</td>
<td>Description</td>
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<td>STC Holder</td>
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<td>Skytec Moveable Class Divider InstallationThis modification introduces a Moveable Class Divider for the installation of an A320 Family aircraft cabin. The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- Equivalent Safety Finding(s): - E-11450301-luminous exit sign for moveable class divider. The following paragraph(s) at a later amendment: CS 25.1, CS 25..2...</td>
<td>- The change is defined with revision A of Skytec Master Document List/Document.MDL/ A32F005-00016. Maintenance shall be accomplished in accordance with revision A of Skytec Instructions for Continuous Airworthiness Document/CAS/ A32FC005-0014. Later revisions of the above listed document(s) approved...</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/12/2023</td>
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<tr>
<td>SKYTEC AEROSPACE GmbH</td>
<td></td>
<td></td>
<td>FANS-1A over Indium Satcom Installation of a Satcom Indium Data Unit with FANS 1A+ capability and removal of the installed Inmarsat Swift 64 satcom system. The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: CS 25.25, CS 25.301, CS 25.303, CS 25.305, CS 25.307, CS 25.601, CS 25.603, CS 25.605, CS 25.607...</td>
<td>- Master Document List/Document.MDL/20A14AE-02 Rev. A- Instructions for Continued Airworthiness ICA20A14AE-02 Rev. A- Flight Manual Supplement/Document.FMS/20A14AE-02 Rev. A- Corresponding changes to the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/12/2023</td>
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<tr>
<td>ALAMO ENGINEERING GmbH</td>
<td></td>
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<td>Zero Pax Cabin on ATR72-212A Removal of all seats for Zero Pax Cabin configuration in preparation of Maintenance activities. The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: CS 25. Artit 27: 25.25, 25.26, 25.29, 25.31, 25.301, 25.303, 25.305 (a) (i), 25.307 (a) (ii), 25.581(a) (b), 25.601...</td>
<td>- AFM supplement/RRC_F0RI01_01_AAL001_04 dated 19th June 2023ICA supplement/RRC_F0RI01_01_AAL001_02 dated 27th November 2023Master Data List: RRC_F0RT5_01_AAL001_00 dated 15th December 2023Cor later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>The airplane must be maintained in accordance with ICA supplement/RRC_F0RI01_01_AAL001_02 and operated in accordance with AFM supplement/RRC_F0RI01_01_AAL001_04. Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other...</td>
<td>19/12/2023</td>
</tr>
<tr>
<td>R&amp;R CONSULTING</td>
<td></td>
<td></td>
<td>Installation of a Viasat KuKa Radome Assembly The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- Equivalent Safety Finding(s): - C-320 vibration/Buffeting Compliance Criteria for Large External Antenna Installation. The following paragraph(s) at a later...</td>
<td>Installation/Viasat KuKa Radome Assembly in accordance with StandardAero Master Data List 1040248 Revision CT The airplane must be maintained in accordance with StandardAero Instructions for Continued Airworthiness 1040243 Revision Cor later revisions of the above listed document(s) approved/accepted...</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/12/2023</td>
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<tr>
<td>STANDARD AERO BUSINESS AVIATION SERVICES ES, L.L.C.</td>
<td>FAA ST0403CH</td>
<td>GIV, GIV-X</td>
<td>Installation of a Viasat KuKa Radome Assembly The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- Equivalent Safety Finding(s): - C-320 vibration/Buffeting Compliance Criteria for Large External Antenna Installation. The following paragraph(s) at a later...</td>
<td>Installation/Viasat KuKa Radome Assembly in accordance with StandardAero Master Data List 1040248 Revision CT The airplane must be maintained in accordance with StandardAero Instructions for Continued Airworthiness 1040243 Revision Cor later revisions of the above listed document(s) approved/accepted...</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/12/2023</td>
</tr>
<tr>
<td>CENTEX AEROSPACE, INC.</td>
<td>FAA ST0904AC</td>
<td>B300, B300C</td>
<td>Halo 250 Conversion Increase of the maximum take-off weight (to 15,950 pounds (7235kg), installation of a take-off in-out-of-range warning system and nose mode stall warning system.</td>
<td>Master Drawing List (MDL) No. 551-0000 Rev A, dated 10 August 2022 Airplane Flight Manual Supplement (AFM5). Document 051-1 Rev IR, dated April 2023&lt;...</td>
<td>Prior to installation of this change/repair it must be determined that the interaction between this change/repair and any other previously installed change/repair and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/12/2023</td>
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<tr>
<td>STC Holder</td>
<td>Description</td>
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<tr>
<td>NSE</td>
<td>Installation of a Lightweight Flight Recorder (LWFR) for data recording and transmission in accordance with STC Data List SDL-00057900AA</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 27 Amendment 10A for 27.1, 27.21, 27.25, 27.27, 27.29, 27.321, 27.31, 27.303, 27.307, 27.561, 27.6...</td>
<td>- Rocketcraft Flight Manual Supplement RMFS-LWFR- EC135 T3 Revision 51- STC Data List SDL-00057900AA Revision A1 - Instructions for Continued Airworthiness ICA-000579000AA Revision A2 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Limited to helicopters with Helionix cockpit variant. The installation of the Iridium Antenna PIN S67-1575-109 according to Ibusbus Helicopters (AH) Illustrated Parts Catalog (IPC) 23.15-02-03 and service connection to enable data acquisition (ARINC429) according to SB EC135H-31-002 are prerequisite.</td>
<td>19/12/2023</td>
<td>Active</td>
</tr>
<tr>
<td>AVIADECO GmbH</td>
<td>12B/C – 164Y/C Passenger Seating Layout This modification introduces a cabin layout generating a 196-passenger seating configuration.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.683 (a), (c), CS 25.1529 (p), CS 25.1729 at Amendment 27. CS-CCD at Issue 2. The requirements for env...</td>
<td>The change is defined with Revision 1 of AviaDeCo Master Data ListMDL-0193C. Maintenance shall be accomplished in accordance with Revision 1 of AviaDeCo Instructions for Continued Airworthiness Document ICA-0193C. The OSD-Cabin Crew Data (CCD) approved per the CS-CCD issue 2 (accompanied by AviaDeCo C).</td>
<td>Prior to installation of this change/repair it must be determined if there is an interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/12/2023</td>
<td>Active</td>
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<tr>
<td>INFINION CERTIFICATION ENGINEERING</td>
<td>Interior Refurbishment</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.1, CS 25.29, CS 25.301, CS 25.561, CS 25.662, CS 25.661, CS 25.603, CS 25.605, CS 25.765, CS 25.769, CS 25.793, CS 2...</td>
<td>The modification is defined in accordance with Infinion Certification Engineering Inc., Master Data List MDX-2727-01, Issue 1 Installation shall be performed in accordance with Infinion Certification Engineering Inc., Installation Instruction EO-2278-01. Issue 1 for later revisions of the above listed document.</td>
<td>This STC is limited to Serial Number 20136 prior to installation of this change/repair it must be determined if there is an interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/12/2023</td>
<td>Active</td>
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<tr>
<td>SAA SOLUTIONS FOR AVIATION, S.L</td>
<td>Installation of Dual Iridium antenna</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The requirements for environmental protection and the associated certified noise and/or emission levels of the product are unchanged and remain applicable to...</td>
<td>Certification Plan Ref. GEX-121-12-30, Edition 1 dated 1st Sep-2023 Maintenance Manual Supplement Ref. GEX-121-12-30, Edition 1 dated 3rd-Nov-2023 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>BD-700-1410 Global Express</td>
<td>22/12/2023</td>
<td>Active</td>
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<tr>
<td>SABENA TECHNICS BOC</td>
<td>Dual cabin reconfiguration with IFE installation Revision 1.2B/C + 26B/C including IFE installation.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 at amdt. 4: 25.1, 25.29, 25.301, 25.303, 25.305, 25.307, 25.321, 25.337, 25.341, 25.365, 25.471, 25.58...</td>
<td>Master Data List, Ref. 41092-01-A-MDL, issue 0 dated 21 December 2023 Instructions for continued airworthiness, Ref. 41092-01-A-ICA, issue 0 dated 21 December 2023 Aircraft Flight Manual Supplement, Ref. 41092-01-A-AMF, issue 0 dated 13 December 2023 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>As a concurrent requirement of this change, accomplishment of this STC requires the simultaneous accomplishment of the following modification: Installation of a non-electrical floor path marking system as per Lufthansa Technik’s EASA STC No. 10014507 (originally EASA A.S. 02580). SIM card shall be...</td>
<td>22/12/2023</td>
<td>Active</td>
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<td>STC Holders</td>
<td>STC Holder</td>
<td>STC Ref</td>
<td>STC Ref</td>
<td>STC Ref</td>
<td>Description</td>
<td>Certification Basis</td>
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<tr>
<td>JANMO AMERICA, INC.</td>
<td>FAA ST024785E</td>
<td>STC Holder</td>
<td>STC Ref</td>
<td>STC Ref</td>
<td>Installation of an interior recouperation/Installation of this STC replaces the low-pressure oxygen flex hose with new non-conductive low-pressure oxygen flex hoses in specific locations and constitutes an approved Alternate Means of Compliance (AMOC) to paragraph (g) of adopted Airworthiness Directives...</td>
<td>The Type and OIS Certification Basis refers to the original product as amended by the following additional or alternative airworthiness requirements: FAA AC25-187A-9C, FAA AC 367-9C, FAA AC 611-9C adopted Equivalent Safety Findings(s): ESF 25.701 as documented in FAA Memorandum...</td>
</tr>
<tr>
<td>JAMCO AMERICA, INC.</td>
<td>FAA ST024785E</td>
<td>STC Holder</td>
<td>STC Ref</td>
<td>STC Ref</td>
<td>Installation of Aerodata ADMS-1100 Mission System</td>
<td>The Type and OIS Certification Basis refers to the original product as amended by the following additional or alternative airworthiness requirements:</td>
</tr>
<tr>
<td>JAMCO AMERICA, INC.</td>
<td>FAA ST024785E</td>
<td>STC Holder</td>
<td>STC Ref</td>
<td>STC Ref</td>
<td>Installation of Aquasat Satcom Software</td>
<td>The Certification Basis refers to the original product as amended by the following additional or alternative airworthiness requirements:</td>
</tr>
<tr>
<td>JAMCO AMERICA, INC.</td>
<td>FAA ST024785E</td>
<td>STC Holder</td>
<td>STC Ref</td>
<td>STC Ref</td>
<td>Installation of Portable ELT(s) change provides instructions to install a Portable ELT in FWD w/forward stowage compartment near Door 1L of the aircraft.</td>
<td>The Type and OIS Certification Basis refers to the original product as amended by the following additional or alternative airworthiness requirements:</td>
</tr>
<tr>
<td>JAMCO AMERICA, INC.</td>
<td>FAA ST024785E</td>
<td>STC Holder</td>
<td>STC Ref</td>
<td>STC Ref</td>
<td>Side Facing Seat Installation</td>
<td>The Type and OIS Certification Basis refers to the original product as amended by the following additional or alternative airworthiness requirements:</td>
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<tr>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>LEE AEROSPACE, INC.</td>
<td>Installation of LAI Windshields</td>
<td>The Certification Basis for the original product remains applicable to the certificate approval. The requirements for environmental protection and the associated certificate noise and/or emissions levels of the product are unchanged and remain applicable to this certificate approval.</td>
<td>Installation Instructions Doc. LAI-INSTL-126, Revision B, dated 11 August 2023(&gt;); Master Data List Doc. LAI- MDL-126 Rev B, dated 11 August 2023(&gt;); Instructions for Continued Airworthiness (CA). Doc. LAI-CA-129, Rev Dated 23 May 2023(&gt;); or later revisions of the above listed document(s).</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product. The STC is valid for aircraft types and models approved...</td>
<td>19/01/2024</td>
<td>Active</td>
</tr>
<tr>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>Installation of a Garmin G500 TXi Autonics Suite and associated System(s). This change includes the installation of a Garmin G500 TXi PFD System, with two displays, in the cockpit and an associated system. Furthermore, a Garmin G730 standby indicator and two GPS/COM/NAV systems.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-23 Amendment 4.23, 29, 251, 301, 303, 305, 307, 337, 561, 601, 603, 605, 607, 609, 611, 613, 625, 671, 771, 777, 1...</td>
<td>Project Description ASR-2022-017, PRD-04-0F/Eight Manual Supplements ASR-2022-017, FMS-05.1-005(S-2) Instructions for Continued Airworthiness ASR-2022-017: 7-CA-11.1-00111-2-000 &gt; Aircraft Flight Manual Supplement, Canard Aerospace Document No. 34-3615-06, Rev. NC, dated 31 January 2017(&gt;); or later revisions of the above listed document(s). approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/01/2024</td>
<td>Active</td>
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<tr>
<td>CANARD AEROSPACE CORPORATION</td>
<td>Upgrade of Collins GLU-520 MRMs and Collins Mode S Transponders in accordance with the FAA STC number 704723CH.</td>
<td>The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements: CS 25 Amendment 4.2, 4.29, 611, 625, 671, 737, 601, 723, 725, 731, 733, 735, 737, 743, 745, 747, 749, 751, 753, 755, 757, 759, 761, 763, 765, 767, 769, 771, 773, 775, 1...</td>
<td>The FAA STC ST04723CH</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product. The STC is valid for aircraft types and models approved...</td>
<td>19/02/2024</td>
<td>Active</td>
</tr>
<tr>
<td>PMV ENGINEERING</td>
<td>ADS-B Out Installation</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 4.2, 4.29, 611, 625, 671, 737, 601, 723, 725, 731, 733, 735, 737, 743, 745, 747, 749, 751, 753, 755, 757, 759, 761, 763, 765, 767, 769, 771, 773, 775, 1...</td>
<td>Master Document List AVI-0238-DML-001</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/01/2024</td>
<td>Active</td>
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<tr>
<td>JET AVIATION AG</td>
<td>WAP4 Activation/Demonstration of aircraft immunity to IEEE 802.11a/b/g/n/ac emissions from PEDs for all flight phases except take-off by means of operation based on EASA CM/ES-003: Activation of existing cabin wireless LAN System provision.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25 Amendment 4.2, 4.29, 611, 625, 671, 737, 601, 723, 725, 731, 733, 735, 737, 743, 745, 747, 749, 751, 753, 755, 757, 759, 761, 763, 765, 767, 769, 771, 773, 775, 1...</td>
<td>Master Data List JBLJ- M-0355BAM-001 ISSUE A: Aircraft Flight Manual Supplement JBLJ-M-0355BAM-001 ISSUE A &gt; Aircraft Flight Manual Supplement JBLJ-M-0355BAM-001 ISSUE A (A) &gt; Aircraft Flight Manual Supplement JBLJ-M-0355BAM-001 ISSUE A (A) &gt; Aircraft Flight Manual Supplement JBLJ-M-0355BAM-001 ISSUE A</td>
<td>This change is applicable to aircraft previously fitted with properly qualified and certified Wireless Access Points(s) meeting the certification basis of the aircraft. Furthermore, the Wireless Access Point(s) installation must be capable to transmit only on IEEE 802.11a/b/g/n/ac protocols, or less...</td>
<td>02/02/2024</td>
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<td>Rev.</td>
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<td>10083783</td>
<td>SIRIUM AEROTECH S.L.</td>
<td>STC-01636</td>
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<td>-</td>
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<td>Rev.</td>
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<td>10083794</td>
<td>3S CERTIFICATION, LLC</td>
<td>FAA STC ST02016WI</td>
<td>0</td>
<td>SEE FAA AML</td>
<td>Installation of the Meggitt Wireless Tire Pressure Sensor</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
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<tr>
<td>10083795</td>
<td>SABENA TECHNICS BGC</td>
<td>REV. 1</td>
<td>AS 332 L, AS 332 L1</td>
<td>AS 332 C, AS 332 C1</td>
<td>TCAS II Installation</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certification/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
</tr>
<tr>
<td>10083796</td>
<td>JADE ENGINEERING PTE. LTD.</td>
<td>A320-251N</td>
<td>Full Economy Class Interior Modification</td>
<td>This design change includes a full-economy-class configuration of ETSEC127A Economy Class seats from Recaro.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>Compliance Document List, JD-41BC03-1 Revision A, 5th Dec 2023 Master Drawing List, JD-41BC02-1 Revision A, 4th Dec 2023 Supplemental ICA and EWS ICA Source Document, JD-41BC01-1 Revision 0, 5th Dec 2023 Aircraft Maintenance Manual Supplement, AMM- AIC418-1, 9th Dec 2023 or later revisions of the above...</td>
</tr>
<tr>
<td>10083800</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>60</td>
<td>DUAL FMS ULS-1Lw Upgrade</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval...</td>
<td>L60X-QCM34-004-M1-ICA Instructions of Continuous Airworthiness Initial Issue, L60X-QCM34-004-M1-AMF Aircraft Flight Manual Supplement Initial Issue, L60X-QCM34-004-M1-MMIS Maintenance Manual Supplement Initial Issue, L60X-QCM34-004 Certification Report Revision 2 or later revisions of the above listed doc...</td>
<td>Applicable to aircraft with the following configuration - Dual Universal ULS-1Lw Flight Management System - Dual Collins Proline 4 Flight Display System. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
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<td>STC Holder</td>
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<td>Foreign/NAA</td>
<td>Model</td>
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<tr>
<td>PEREGRINE LLC</td>
<td>FAA STC</td>
<td>ST0048W</td>
<td>SEE FAA AML</td>
<td>787-8</td>
<td>Boeing 787-8 - Avianca Cabin Reconfiguration - 250 to 291 PA/3 to 291</td>
<td>Installation of the Garmin GI-275 Instantaneous Vertical Speed Indicators.</td>
</tr>
<tr>
<td>PEREGRINE LLC</td>
<td>FAA STC</td>
<td>ST0198W</td>
<td>SEE FAA AML</td>
<td>787-8</td>
<td>Boeing 787-8 - Avianca Cabin Reconfiguration - 250 to 291 PA/3 to 291</td>
<td>Installation of the Garmin GI-275 Instantaneous Vertical Speed Indicators.</td>
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<td>DASSAULT AIRCRAFT SERVICES (DAS)</td>
<td>FAA STC</td>
<td>ST0314NY</td>
<td>FALCON 900EX</td>
<td>787-8</td>
<td>Boeing 787-8 - Avianca Cabin Reconfiguration - 250 to 291 PA/3 to 291</td>
<td>Installation of the Viasat Global Aero Terminal GAT-5610 Ka-Band Satellite Communications System.</td>
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<tr>
<td>BOEING IRELAND LIMITED</td>
<td>787-8</td>
<td></td>
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<td></td>
<td>Boeing 787-8 - Avianca Cabin Reconfiguration - 250 to 291 PA/3 to 291</td>
<td>The Type and OSD Certification Bases (CB) for the original product and associated technical documents applicable to this certification approval.</td>
</tr>
<tr>
<td>AEROTEC &amp; CONCEPT</td>
<td>SEE APPROVED MODEL LIST (AML)</td>
<td>LTE TRANSCIEVER INSTALLATION</td>
<td></td>
<td></td>
<td>Boeing 787-8 - Avianca Cabin Reconfiguration - 250 to 291 PA/3 to 291</td>
<td>The Type and OSD Certification Bases (CB) for the original product and associated technical documents applicable to this certification approval.</td>
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**List of EASA Supplemental Type Certificates**

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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10083873</td>
<td>DUKE AVIATION CO. LLC</td>
<td>FAA SA00153DE</td>
<td>60, A60, B60</td>
<td>Installation of 130 Ampere alternator retrofit system to replace existing generator system</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: - The following paragraph(s) at a later amendment: CS 23.1385(e)(1)(2). The requirements for environmental protection and the associated certificated noise levels.</td>
<td>Firewall AERONET Installation Data Package, Revision B, dated March 14, 1995Flight Manual Supplement, document PWK-FLMS-Rev no, dated April 12, 1995 for later revisions of the above listed document(s) approved/ accepted in behalf of EASA in accordance with the Technical Implementation Procedures of E...</td>
<td>Prior to installation of this change/repair it must be determined that the interpersonal relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/02/2024</td>
<td>Active</td>
<td>10083873</td>
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<tr>
<td>1008383</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>A300F4-605ST</td>
<td></td>
<td>Special transport – 2nd Airbus Helicopter H160 on MPPSThis approval covers Issue B of ATI modification ST25MOD020 for the transport of two Airbus Helicopter H160 in the Beluga ST Main Deck Cargo/CompartmentsIssue 1 of ATI modification ST25MOD020 (applicant’s reference) was developed, reviewed, tested as...</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise level remain applicable to this certificate/approval w. - Change Approval Sheet (CAS) ST25MOD020Ed59 Issue 2 dated 09 February 2024- Change Certification Plan (CCP) ST25MOD020Ed59 Issue 4 dated 12 January 2024- Approved Document Dossier (ADD) ST25MOD020 Ed59 ADD Issue 3 including AFM, MM and MMEL supplements as included in Independent Check SheetICS002 Issu...</td>
<td>- The dangerous goods installed in the payloads have to comply with TheATA Dangerous Goods Regulation (or any relevant Aviation Authority regulation). - This approval covers H160 helicopter as described in Airbus Helicopter Technical Notes UST954009801 Issue B and complemented by preparation instruction...</td>
<td>09/02/2024</td>
<td>Active</td>
<td>1008383</td>
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<tr>
<td>1008387</td>
<td>TURK HAVA YOLLARI TEKNIK A.S. (TURKISH TECHNIC Inc.)</td>
<td>A321-231</td>
<td></td>
<td>220Y/C Passenger Seating Layout: This modification introduces a cabin layout generating a 220 passenger seating configuration.</td>
<td>The change is defined with Revision 5 of Türk Hava Yolları Teknizertifikat Programme C0309- CEP-05&lt;,&gt;()or later revisions of the above listed document(s) approved/approached under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interpersonal relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/02/2024</td>
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<tr>
<td>1008389</td>
<td>ALAMO ENGINEERING GmbH</td>
<td>737-700, 737-800</td>
<td></td>
<td>Installation of Ku-SAT SD Plane SimpliBasin on existing EASA approvals for a sat-in radome and an aromatic internal date(1)(2) voice communication network (including wireless and B-Band Satcom) on a Ku-Band Satcom antenna system has been installed here. The antennas and microwave unit were mounted in th...</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certificated noise level remain applicable to this certificate/approval w. - Instructions for Continued Airworthiness ICA21AS50AE-01 Rev A/Flight Manual Supplement FMES21AS50AE-01 Rev A Master Document List (MDL) MLD21A50AE-01 Rev A(1)(2)or later revisions of the above listed document(s) approved/approached under the EASA system.</td>
<td>EASA STC No. 1006445, Luftfrans Technik Two-In-One-Automation (TODS) maybe previously or concurrently embodied for the aim of the present modification. Prior to installation of the change/repair it must be determined that the interpersonal relationship between this change/repair and any other previously installed...</td>
<td>13/02/2024</td>
<td>Active</td>
<td>1008389</td>
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<tr>
<td>10083907</td>
<td>COMPANIA ESPANOLA DE TECNOLOGIA AVIONICA UTICA, S.A.</td>
<td>1900D</td>
<td></td>
<td>Installation of Dual GTX300 Transponder for ADS-B Out capability</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: Refer to Amendment dated 15/02/2004. The requirements for environmental protection and the associated certificated noise...</td>
<td>Master Document and Drawing List, document D2305-0001, Ed 02Aircraft Flight Manual Supplement, document D2305-0017, Ed 01Maintenance Manual Supplement, document D2305-0018, Ed 01or later revisions of the above listed document(s) approved/approached under the EASA system.</td>
<td>Applicable to Textron Beechcraft aircraft Model 1900D S/N: UE-107, UE-69, UE-301, UE-387, UE-248, UE-103, UE-196 with the following pre-modification configuration as detailed in Modification Description ref. D2305-0003-03. Prior to installation of this change/repair it must be determined that the interpersonal relationship...</td>
<td>15/02/2024</td>
<td>Active</td>
<td>10083907</td>
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<td>EASA STC No.</td>
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<td>FAA STC/SAO/STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>10083908</td>
<td>TEXTRON AVIATION Inc.</td>
<td>FAA STC SAU0486WI</td>
<td>1900D</td>
<td>Installation of Cabin Ordinance (Fasten Seat Belt / No Smoking) Signs in Beech Model 1900D Airplanes.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Raytheon Aircraft Services Master Drawing List 16-000039-MDL, Revision (→), dated 21 November 1996, or later revisions of the above list document(s) approved/accepted on behalf of FAA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2024</td>
<td>Active</td>
<td>10083910</td>
<td></td>
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<tr>
<td>10083910</td>
<td>BELLTEXTRON, Inc.</td>
<td>FAA STC SR11252GS</td>
<td>412EP</td>
<td>Installation of Additional Host Cable Cut Switches</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>1. Master Drawing List, Report No. AA-220233, Revision B, dated 13 April 2023.2. Aeronautical Accessories Rotorcraft Flight Manual Supplement, Report No. AA-220391(→), No Revision, dated 7 August 2023.3. Aeronautical Accessories Instructions for Continued Airworthiness (→), Report No. AA-22024, Nov...</td>
<td>This STC is applicable to all serial numbers for Bell Textron Inc. Model 412EP. Installation of FAA STC SH3189SO (External Non-HEC Helis) is approved on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>27/06/2024</td>
<td>Active</td>
<td>10083910</td>
<td></td>
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<tr>
<td>10083914</td>
<td>NEXTANT AEROSPACE, L.L.C.</td>
<td>FAA STC ST04612CH</td>
<td>BD-700-1A10</td>
<td>Installation of a SpaceX Starlink Avation System</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Nextant Aerospace Master Data List, Doc. No. NT0079-MDL-0100(→), dated 30 January 2024.1. Airplane Flight Manual Supplement, Nextant Aerospace Doc. No. NT0079-AMS-0100(→), Revision (→), approved 2 February 2024.2. Instructions for Continued Airworthiness – Airworthiness Limitations Chapter...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. This modification requires that the aircraft be shown t...</td>
<td>16/02/2024</td>
<td>Active</td>
<td>10083914</td>
<td></td>
</tr>
<tr>
<td>10083918</td>
<td>PERERGRINE LLC</td>
<td>FAA STC SAU0331Y1</td>
<td>AT-802A AT-802</td>
<td>AT-602</td>
<td>Installation of a True Blue Power TB60 Lithium Ion Battery to replace the existing lead-acid battery</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>Master Drawing List (MDL), No. E-WP-22-0002, Rev. IR, dated 24 January 2024(→), Airplane Flight Manual Supplement (AFMS), No. E-WP-22-0009, Rev. IR, dated 24 January 2024(→), &gt;or later revisions of the above list document(s) approved/accepted on behalf of EASA in accordance with the Technical Impl...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2024</td>
<td>Active</td>
<td>10083918</td>
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<tr>
<td>10083919</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST04349AT-D</td>
<td>GV-SP GV-X</td>
<td>Laserf V to Laserf VI Upgrade</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>1. Gulfstream Aerospace Corporation (Index List GDL 100000077, Revision(s)), dated 2 November 2023.2. FAA accepted Instructions for Continued Airworthiness CE41113AD02, Revision(s) dated 11 October 2023 for Gulfstream Model GIV-X.3. FAA accepted Instructions for Continued Airworthiness CE41113AD02...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2024</td>
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<td>10083919</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NA</td>
<td>3TC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Certification</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>10083933</td>
<td>MAGELLAN AEROSPACE</td>
<td>REV. 1</td>
<td>TCCA SH93-41</td>
<td>212</td>
<td>Installation of Wire Strike Protection System.</td>
<td>The Type and OSD Certification Bases (CB) for the original products remain applicable to this certificate/approval. The requirements for environmental protection and the associated/designated noise and/or emissions levels of the product are unchanged and remain applicable.</td>
<td>Configuration: Installation P/N 384-83300-0 (Standard Sdtd Gear) or P/N 384-83300-3 (High Sdtd Gear) + Drawing List EM105301 Issue F TCCA Approved 30 June 2011.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2024</td>
<td>Active</td>
<td>10053023</td>
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<tr>
<td>10083940</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>A320-232</td>
<td>TCCA G815-5</td>
<td>GULFSTREAM 200/ GALAXY</td>
<td>180 YC Passenger Seating Layout: The modification introduces a cabin layout generating a 180-paxpassengereo seating configuration.</td>
<td>The Type and OSD Certification Bases (CB) for the original products remain applicable to this certificate/approval. The requirements for environmental protection and the associated/designated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Configuration: Installation P/N 384-83100-11.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/02/2024</td>
<td>Active</td>
<td>10062040</td>
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<tr>
<td>10083963</td>
<td>SKY SERVICE F.B.O. INC.</td>
<td>TCCA GA15-5</td>
<td>Installation of a WiFi system with FANS-1/ A+ Data Link Option.</td>
<td>Dual UASC UMS-16iEw FMS with FANS-1/ A+ Data Link Option.</td>
<td>The Type and OSD Certification Bases (CB) for the original products remain applicable to this certificate/approval. The requirements for environmental protection and the associated/designated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Configuration: Installation P/N 384-83100-11.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/02/2024</td>
<td>Active</td>
<td>10062063</td>
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<tr>
<td>10083965</td>
<td>STELLAR BLU SOLUTIONS GmbH</td>
<td>A319-153N/-171N</td>
<td>A319-1311/-1321/-1331/-151N</td>
<td>CABANNER Infotainment System Installation/Installation of a WiFi system and LCD screen on passenger seats.</td>
<td>The Type and OSD Certification Bases (CB) for the original products remain applicable to this certificate/approval. The requirements for environmental protection and the associated/designated noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Configuration: Installation P/N 384-83100-11.</td>
<td>Installation on aircraft models only that satisfy the configurations and criteria defined in document FCS-233- A319 Iss.01, or later revisions approved under the EASA system. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/02/2024</td>
<td>Active</td>
<td>10062065</td>
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<tr>
<td>10083987</td>
<td>ICELANDAIR EHF</td>
<td>767-300</td>
<td>Structural provisions for SATCOM system: This modification introduces structural provisions for a SATCOM system for 767-300 BCF aircraft.</td>
<td>Structural provisions for SATCOM system: This modification introduces structural provisions for a SATCOM system.</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended.</td>
<td>Configuration: Installation P/N 384-83100-11.</td>
<td>Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/02/2024</td>
<td>Active</td>
<td>10062067</td>
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<td>STC Holder</td>
<td>Description</td>
<td>Certification Basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>DIAMONDO AIRCRAFT INDUSTRIES GmbH</td>
<td>MTO 2360 kg (to increase the operational range/insurance of the DA 62) - Maximum Take-Off Mass will be increased to 2360 kg. The Maximum Landing Mass and/or Maximum Zero Fuel Mass will not be changed.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Installation per STC 62-005 and/or AFM Supplement STC-62-005 Rev. 0 to Doc. No. 7.01.25-E/11.01.05-E and AFM Supplement STC-62-005 Rev. 0 to Doc. No. 7.01.25-E/11.01.05-E of later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/02/2024</td>
<td>Active</td>
<td>10082990</td>
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<tr>
<td>JET AVIATION AG</td>
<td>Seat to floor interface</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List, Ref. JBSL-M-317/MDL-001 issue B dated 24 November 2023 and/or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/02/2024</td>
<td>Active</td>
<td>10082995</td>
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<tr>
<td>AIRCRAFT INTERIOR REFURBISHMENT ESPANA S.L.</td>
<td>Rearrangement and installation of passenger seats.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List, Ref. A330C16-MDL-001 issue 2 dated 30 January 2004 instructions for continued airworthiness, Ref. A330C16/MM/001 issue dated 25 October 2023 and/or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/02/2024</td>
<td>Active</td>
<td>10084002</td>
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<tr>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>Installation of Honeywell AIS-2000 High Definition (HD) Upgrade on Gulfstream Models: G-IV, GIV-X, GV, GV-SP and GV1.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval.</td>
<td>For Gulfstream Aerospace Model G-IV (&gt;,&lt;) For Gulfstream Aerospace Model GV-X, GV, GV-SP and GV1.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/03/2024</td>
<td>Active</td>
<td>10084007</td>
<td></td>
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<tr>
<td>CENTRI-LUBE TECHNOLOGIES, LLC</td>
<td>Engine baffle access panel installation.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Firewall Forward Drawing No. FFT-PRMR-0001, Rev. dated 7/9/2023 and/or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/02/2024</td>
<td>Active</td>
<td>10084015</td>
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<tr>
<td>J &amp; C AEROSPACE DESIGN AND CERTIFICATION SERVICES EUU Ltd</td>
<td>3D model:- Introduction of single class economy cabin RaceAir reconfiguration from current configuration to single-class cabin to single-class by placement of existing ETOSO c127 with new ETOSO c127a on GIV-X model</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval.</td>
<td>Master Data List ref: JCAERO/MLTI/178 issue 1 dated 1st May 2024 and/or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Aeroplanes operating with this STC must be maintained in accordance with the applicable Airworthiness Information Document (AID) of the STC-62-005.</td>
<td>01/03/2024</td>
<td>Active</td>
<td>10084024</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC no.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>10084032</td>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td>FAA STC</td>
<td>ST04483AT</td>
<td>BD-500-1A11, BD-500-1A10</td>
<td>Installation of the Avions Support Group Constant Friction Mounting Mount with Cruise</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: the following paragraph(s) at a later amendment: CS25 0: 25.29, 25.303, 25.305, 25.306, 25.307, 25.773, 25.789, 25.1529, 25.1541C5 25 amd. 8, 25.305, 25.307, 25.613C...</td>
<td>1. Master Drawing List, 219223-MDL, Revision A, dated 14 January 2020; 2. Instructions for Continued Airworthiness (ICA) document 219223-ICA, Rev. IF, dated 31 October 2019; or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementing...</td>
<td>Prior to installation of this change/repair it must be determined that this change/repair and any other previously installed change/repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>29/04/2024</td>
<td>Active</td>
<td>10084032</td>
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<tr>
<td>10084056</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC</td>
<td>SA04033E</td>
<td>AT-501, -502, -502A, -502B, AT-401B, -402B, AT-400, -400A, -401, -401A, AT-300-301-302, AT-203, -300, -400</td>
<td>Installation of Vortex Generators on the wings</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>MicroAerodynamics Drawing Package MA2073, Revision C, dated 10 March 2013; for Type Certificate (TC) ASWV aircrafts/MicroAerodynamics Drawing Package MA2074, Revision C, dated 19 March 2013, for TC A17SW aircrafts; and Installation Manual MA2075, Revision D, dated 19 March 2013, or later revisions; etc...</td>
<td>Prior to installation of this change/repair it must be determined that this change/repair and any other previously installed change/repair will introduce no adverse affect upon the airworthiness of the product. If more than five vortex generators are damaged or...</td>
<td>05/03/2024</td>
<td>Active</td>
<td>10084056</td>
</tr>
<tr>
<td>10084057</td>
<td>LUFTFABENS KO (AMMS)</td>
<td>FAA STC</td>
<td>SA04033E</td>
<td>A320-214</td>
<td>Installation of Vortex Generators on the wings</td>
<td>The Type and OSD Certification Bases (CB) for the original product remain applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Only applicable to aeroplanes not equipped with JAR25.1582 unless the...</td>
<td>Prior to installation of this change/repair it must be determined that this change/repair and any other previously installed change/repair will introduce no adverse affect upon the airworthiness of the product. If more than five vortex generators are damaged or...</td>
<td>06/03/2024</td>
<td>Active</td>
<td>10084057</td>
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<tr>
<td>10084071</td>
<td>ANIVU OPERATIONS LLC</td>
<td>FAA STC</td>
<td>ST14930AT</td>
<td>737-8</td>
<td>Installation of Row 44 Broadband Antenna provisions on Boeing 737-8</td>
<td>The Certification Basis for the original product as amended the following additional or alternative airworthiness requirements: CS25 0: 25.29, 25.31, 25.21, 25.23, 25.27, 25.111, 25.113, 25.119, 25.121, 25.13, 25.177, 25.181, 25.21, 25.301, 25.303, 25.305, 25.307, 25.331, 25.335, 25.341, 25.349, 25.3...</td>
<td>TIMCO Aerosystems, LLC Master Data List 17T062LO01, Revision K, dated January 2020; &gt;TIMCO Aerosystems Manual Supplement A-25-23/A FM 1 Issue 1 Rev. 0</td>
<td>Prior to installation of this change/repair it must be determined that this change/repair and any other previously installed change/repair will introduce no adverse affect upon the airworthiness of the product.</td>
<td>29/04/2024</td>
<td>Active</td>
<td>10084071</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Disclaimment</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>10084081</td>
<td>RECAERO</td>
<td></td>
<td>A330-202</td>
<td></td>
<td>Cabin reconfiguration</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following special condition(s): CS-ACNS. B.V., for installation of Garmin G500H TXi Flight Deck into MD Helicopters MD 902.</td>
<td>Master Document List (MDL) Doc No. RCP036MDL.001 Issue Rev. 1 Operating Manual Supplement Doc No. RCP036AMS.001 Issue Rev. 1 Operating Manual Supplement Doc No. RCP036AMS.001 Issue Rev. 1 for later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Installation is not NVIS approved. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>11/03/2024</td>
<td>Active</td>
<td>10084081</td>
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<tr>
<td>10084086</td>
<td>SPAES GmbH &amp; Co. KG</td>
<td></td>
<td>EC135 T2+ EC135 T1, EC135 T2 EC135 P1, EC135 P3, EC135 P2+</td>
<td></td>
<td>Installation of an external VHF Antenna in accordance with MasterDocument List MDL-135-660-23-01</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following special condition(s): CS-ACNS. B.V., for installation of an external VHF Antenna in accordance with MasterDocument List MDL-135-660-23-01.</td>
<td>- Rotorcraft Flight Manual Supplement FMS-135-660-23-01 Revision 1.0- Master Document List MDL-135-660-23-01 Revision 5 - Instrucions for Continued Airworthiness ICA-135-660-23-01 Revision 2 - Operating Manual Supplement Doc No. RCP036AMS.001 Issue Rev. 1 for later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Limited to helicopters with CP/OS (co)ukt variant/kit/ retrofit of the emergency floatation system as per Airbus Helicopters Service Bulletin (SB) EC135-32-011 it is prerequisite. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>12/03/2024</td>
<td>Active</td>
<td>10084086</td>
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<tr>
<td>10084109</td>
<td>HELI AUSTRALIA GmbH</td>
<td></td>
<td>MD900</td>
<td></td>
<td>Installation of Garmin G500H TXi Flight Deck into MD Helicopters MD 902</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following special condition(s): CS-ACNS. B.V., for installation of Garmin G500H TXi Flight Deck into MD Helicopters MD 902.</td>
<td>Master Document List, Doc No. HAU-D-065-001, Rev. 0 Operating Manual, Doc No. HAU-D-065-004, Rev. 0 Operating Manual Supplement, Doc No. HAU-D-065-006, Rev. 0 Operating Manual Supplement, Doc No. HAU-D-065-008, Rev. 0 for later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Installation of these items do not introduce any adverse affect upon the airworthiness of the product.</td>
<td>14/03/2024</td>
<td>Active</td>
<td>10084109</td>
</tr>
<tr>
<td>10084111</td>
<td>KNIS AVIATION ENGINEERING LIMITED</td>
<td>REV. 1</td>
<td>UK STC.00070 REVISION 00</td>
<td>A321-251NX</td>
<td>Installation of ACRQ924600 ICA-135-924600-23-01 (E) Economy Class Seats (232 pax)</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following special condition(s): CS-ACNS. B.V., for installation of ACRQ924600 ICA-135-924600-23-01 (E) Economy Class Seats (232 pax).</td>
<td>Master Document List, Doc No. HAU-D-065-001, Rev. 0 Operating Manual, Doc No. HAU-D-065-004, Rev. 0 Operating Manual Supplement, Doc No. HAU-D-065-006, Rev. 0 Operating Manual Supplement, Doc No. HAU-D-065-008, Rev. 0 Operating Manual Supplement, Doc No. HAU-D-065-008, Rev. 0 for later revisions of the above listed document(s) approved/accepted/under the EASA system.</td>
<td>Limited to VFR operation. Installation is not NVIS approved. Prior to installation of this change/repair it must be determined that there is no adverse effect upon the airworthiness of the product.</td>
<td>18/03/2024</td>
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<td>10084111</td>
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<td>Doc. No.</td>
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<tr>
<td>STC-180-G-II</td>
<td>SIA “AEROKNOW”</td>
<td>Installation of a Low-Frequency Underwater Locator Beacon (LF-ULB)</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: JAR-25 Change 15(Sections 25.29, 25.301, 25.303, 25.305, 25.307, 25.321, 25.337, 25.341,25.981, 25.571, 25.8...</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/03/2024</td>
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<td>10084123</td>
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<td>STC-180-G-AMM</td>
<td>ICARUS AEROTECHNICS</td>
<td>STC ICARUS 006 - PC-6 Reinforced Flap</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/03/2024</td>
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<td>10084125</td>
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<td>STC-180-G-AFM</td>
<td>DUNCAN AVIATION, Inc.</td>
<td>Installation of Norwegian Special Mission UNIFIS 3000 Flight Inspection System</td>
<td>(1) Duncan Aviation, Inc., STC Master Document List CERTDOC-000001863, Rev. B dated 17 June 2015&lt;sup&gt;1&lt;/sup&gt;</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/03/2024</td>
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<td>10084129</td>
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<td>STC-180-G-AFM</td>
<td>HANSJÖRG STREIFENEDER FLUGFASE- FLUZIEG SERVICE GmbH</td>
<td>Hand rudder control for ASH 25</td>
<td>- Flight manual supplement: Anhang A Flughandbuch Sitz- und Standsteuerung, Ausgabe 01, Stand 03.05.2023. Maintenance manual supplement: Anhang B Wartungshandbuch, Ausgabe 01, Stand 03.05.2023.</td>
<td>Self-launch with hand rudder control is not permitted. Prior installation of this change/repair must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/03/2024</td>
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<td>10084132</td>
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<td>Description</td>
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<td>10084162</td>
<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>ST00202B0</td>
<td>BD-700-1A11/1A0</td>
<td>Installation of Environmental Control System Ice Pellet Filler</td>
<td>The Certification Basis for the original product remains applicable with certificate approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Bombardier Aerospace Master Document List BAA87107014/14 (Rev.B, dated 6 April 2005) (x); Instructions for Continued Airworthiness BAA87107014-FC-1A, Rev.A, dated 7 April 2005 (x); Latest revision of the above listed document(s) approved/applicable on behalf of EASA in accordance with the Technics.</td>
<td>Prior to installation of this change/repair it must be determined that there is no adverse interaction between this change/repair and any previously installed change and/or repair. Will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/03/2024</td>
<td>Active</td>
<td>10084162</td>
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<tr>
<td>10084187</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>PA-46-500TP</td>
<td></td>
<td></td>
<td>Installation of Pratt &amp; Whitney PT6A-135A engine in PA-46-500TP aircrafts.Pre-requisites for the Installation of the PT6A-135A engine are: - MT-Propeller STC EASA 10047226 / FAA SA03446NY (Installation of Blade MT-Propeller Model MTV-27-1-E-C-F-R(P)/CFR203-6A)).</td>
<td>The Type and OSD Certification Bases (CB) for the original propellers applicable to this certificate/approval involves a change with an impact on the noise database and/or emission levels. The environmental protection requirements are amended as follows: CS-36 AMDt 6, FAA.</td>
<td>According to MT-Propeller Master Document List No. E-3933 Rev. 0 (To be installed in accordance with MT-Propeller AFM-S Document No. E-3932Rev. 0 To be installed i.a.w. MT-Propeller ICA Doc. No. E-3954 Rev. 0, later...</td>
<td>1. According limitation section of AFM-S. Pre-requisites for the installation of the PT6A-135A engine are: - MT-Propeller STC EASA 10047226 / FAA SA03446NY (Installation of Blade MT-Propeller Model MTV-27-1-E-C-F-R(P)/CFR203-6A)) - Garmin STC 10064794 / FAA SA02571E</td>
<td>25/03/2024</td>
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<tr>
<td>10084189</td>
<td>PMV ENGINEERING</td>
<td>BD-500-1A10</td>
<td></td>
<td></td>
<td>CABIN HUMIDIFICATION SYSTEM<em>STEP 1</em> PROVISIONS</td>
<td>The Type and OSD Certification Bases (CB) for the original propellers applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w.</td>
<td>Master Data List “Provisions” ECS-00130-MDL-001 Rev 012A (supplement ECS-00130-ICA Rev 00r later revisions of the above listed document(s) approved/applicable under the EASA system.</td>
<td>- This certificate does not cover the activation of the humidification-system. The MBBR and COPP data must be delivered within 6 months (i.e. Step 2) after issuance of this certificate/approval.</td>
<td>26/03/2024</td>
<td>Active</td>
<td>10084189</td>
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<tr>
<td>10084198</td>
<td>DOA21 LIMITED</td>
<td>REV. 1</td>
<td>BD-500-1A10</td>
<td></td>
<td>Introduction of cabin Interior “ATT2”Rev. 1 TYPE connector limitations section.</td>
<td>The Type and OSD Certification Bases (CB) for the original propellers applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Master Document List D21-MDL-0007 Issue 1AFM Supplement D21-FMS-0002 Issue 2Maintenance Planning Document D21-MPS-0004 Issue 10OSD FOO crew training D21-FOOTR-0002 Issue 20OSD CDDAircraft Difference Table D21-ADT-0002 Issue 20OSD MMEL Supplement D21_MME-0003 Issue 3r later revisions of the above...</td>
<td>- Airworthiness limitations as per Maintenance Planning Document. Aeronautical configurations changed per this STC are considered to be a “variant” to the A220 (BD500-100/-300) commercial configurations. Cabincrew training on aircraft types changeper this STC must complete aircraft/reference training in ac...</td>
<td>03/04/2024</td>
<td>Active</td>
<td>10084198</td>
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<td>10084223</td>
<td>HONEYWELL INTERNATIONAL Inc.</td>
<td>FAA STC 076SLA</td>
<td>BD-100-1A10</td>
<td></td>
<td>Installation of Honeywell EDG-100 Connected Engine Data Access System</td>
<td>The Type and OSD Certification Bases (CB) for the original propellers applicable to this certificate/approval. Additional or alternative environmental protection requirements are amended as follows: CS-36 AMDt 6, FAA.</td>
<td>Installation of the Honeywell EDG-100 Connected Engine DataAccessSystem in accordance with Honeywell Master Drawing List 756-0275-001, Revision A.Honeywell Instructions for Continued Airworthiness document number756-0375-850, Revision A must be made available to the operator at the latest revision of the...</td>
<td>The installer must determine that the A9307-1-1A Light Maintenance Manual, 7.05-22 supports the use of the EDG-100 Connected Engine DataAccessSystem prior to incorporating this modification on a Challenger350. The installer must determine that the A9307-1-1A Light Maintenance Manual, 7.05-12 support...</td>
<td>02/04/2024</td>
<td>Active</td>
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<td>STC Holder</td>
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<td>DASSAULT AIRCRAFT SERVICES (DA6)</td>
<td>MYSTERE FALCON 900 (IMF900) + FALCON 900EX (F900EX)</td>
<td>10084242: ESA-0025-0005 Rev. D. AFMS F9XW- D0103-150 Rev. A - ICA F9XW-D0103-155 Rev. B re-later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>D2318-0001 Edition 03Rotorcraft Flight Manual Supplement, Rev. 04/2024. The above listed document(s) approved/accepted on behalf of EASA.</td>
<td>05/04/2024: Active</td>
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<td>10084242</td>
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<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td>Boeing 737-800 - Cabin Reconfiguration (189 Y/C) this project installs a 42 galley at the forward LH area and reconfigures the current 189 Y/C LOPA into a new 189 Y/C LOPA.</td>
<td>737-800: ESA-0025-0005 Rev. D. AFMS F9XW- D0103-150 Rev. A - ICA F9XW-D0103-155 Rev. B re-later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>D2318-0014 Edition 03Rotorcraft Flight Manual Supplement, Rev. 04/2024. The above listed document(s) approved/accepted on behalf of EASA.</td>
<td>05/04/2024: Active</td>
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<td>10084243</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>A330-203: Incomplete cabin (MSN 655): Removal of all seats except one row of 8 seats and CASS at Door 1 and Door 2; all monuments except the Galley/01 and Lavatories L1 L7 L8 Rev. 1 corrects typographical errors.</td>
<td>10084244: ESA-0025-0005 Rev. D. AFMS F9XW- D0103-150 Rev. A - ICA F9XW-D0103-155 Rev. B re-later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>D2318-0001 Edition 03Rotorcraft Flight Manual Supplement, Rev. 04/2024. The above listed document(s) approved/accepted on behalf of EASA.</td>
<td>05/04/2024: Active</td>
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<tr>
<td>COMPANIA ESPANOLA DE TECNOCAERONAUTICA</td>
<td>Installation of the airborne mission system for Search and Rescue (SAR) Lifesaver SAR XL S10 in accordance with Master Document and Drawing.dsd2318-0001</td>
<td>10084245: ESA-0025-0005 Rev. D. AFMS F9XW- D0103-150 Rev. A - ICA F9XW-D0103-155 Rev. B re-later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>D2318-0001 Edition 03Rotorcraft Flight Manual Supplement, Rev. 04/2024. The above listed document(s) approved/accepted on behalf of EASA.</td>
<td>05/04/2024: Active</td>
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<td>LUFTHANSA TECHNIS AG</td>
<td>A320: Door Lock Installation/External and internal door locking devices.</td>
<td>10084246: ESA-0025-0005 Rev. D. AFMS F9XW- D0103-150 Rev. A - ICA F9XW-D0103-155 Rev. B re-later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>D2318-0001 Edition 03Rotorcraft Flight Manual Supplement, Rev. 04/2024. The above listed document(s) approved/accepted on behalf of EASA.</td>
<td>05/04/2024: Active</td>
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<td>FLYHT AEROSPACE SOLUTIONS LTD.</td>
<td>Installation of the AFIRS 228 System</td>
<td>10084249: ESA-0025-0005 Rev. D. AFMS F9XW- D0103-150 Rev. A - ICA F9XW-D0103-155 Rev. B re-later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU USA Bilateral Agreement.</td>
<td>D2318-0001 Edition 03Rotorcraft Flight Manual Supplement, Rev. 04/2024. The above listed document(s) approved/accepted on behalf of EASA.</td>
<td>05/04/2024: Active</td>
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<td>10084249</td>
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<td>10084257</td>
<td>NOVA SYSTEMS &amp; ENGINEERING LIMITED</td>
<td></td>
<td>A321-231, A321-232, A321-211, A321-212, A321-213</td>
<td>Airbus A321-200 – Cabin Interior Reconfiguration</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-25D Issue 5. The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Master Data List “Airsas A321-200 – Cabin Interior Reconfiguration” NIS-20861 0-01-MDL-01, Issue 1, dated 8th April 2024&lt;sup&gt;1&lt;/sup&gt; or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>This STC is not applicable to A321-211, A321-212, A321-213, A321-211, andA321-221 if they have any of the following Airbus modifications/Mod: MOD 157272 Max Pax MOD 159538 Max Pax MOD 160139 Passenger information signs and placards Prior to installation of this change/repair it must be determined if...</td>
<td>09/04/2024</td>
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<td>10084272</td>
<td>AERODATA AG</td>
<td>680A (LATITUDE)</td>
<td>Installation of Aerodata AG Flight Inspection System AD-AFIS-1320-Aerodata Modification Ident: OB80A-02</td>
<td>Installation of Aerodata AG Flight Inspection System AD-AFIS-1320-Aerodata Modification Ident: OB80A-02</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-25D Issue 5. The requirements for environmental protection and the associated certified noise and/or emissions...</td>
<td>Master Document List Installation of AD-AFIS-1320 in C680A document MDL-A/C-971 Rev 0, dated 27 March 2024 Airplane Flight Manual Supplement Installation of AD-AFIS-1320 in C680A document AFM-191 Rev 0, dated 16 January 2024 Aircraft Maintenance Manual Supplement Installation of AD-AFIS-1320 in C680A M...</td>
<td>The performance of Aerodata Flight Inspection System AD-AFIS-1320 hasn't been assessed by EASA. This STC cannot apply to aircaft configurations where any flight crewmembers have been added in addition to the ones implemented in theoretical TC design and are not integrated with the original TC Cre...</td>
<td>11/04/2024</td>
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<td>10084276</td>
<td>Q.C.M. DESIGN GmbH</td>
<td>525</td>
<td>Collins Indium Certus Satcom Installation</td>
<td>Collins Indium Certus Satcom Installation</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged/remain applicable to this certificate/ approval w...</td>
<td>Engineering Bulletin ref EB-CS25-02M2-001 Rev 0; Instructions for Continuous Airworthiness Ref CS25-02M2-001-M1-ICA Rev 0 Flight Manual Supplement Ref CS25-02M2-001-M1-ICA Rev 0 Certificate Report Rev 0; later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined whether the change/repair and any other previsously installed change and/or repair will introduce no adverse effect upon the aeroworthiness of the product.</td>
<td>11/04/2024</td>
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<td>10084286</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>A320-232</td>
<td>Cabin reconfiguration to 180 pax</td>
<td>Installation of UASC LPV Monitor for PBN</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: FAR-25 Amendment 25-136; 25-29; 25-251; 25-301; 25-305; 25-307; 25-337; 25-341; 25-365; 25-561; 25-57...</td>
<td>Master Data List AE23088MDL Issue 0; Instructions For Continued Airworthiness AE23088CA Issue 0; later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined whether the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the aeroworthiness of the product.</td>
<td>12/04/2024</td>
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<td>10084291</td>
<td>COMPANIA ESPANOLA DE TECNOLOGIA AERONAUTICA, S.A.</td>
<td>550</td>
<td>Installation of UASC LPV Monitor for PBN</td>
<td>Installation of UASC LPV Monitor for PBN</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: FAR-25 Amendment 25-136; 25-29; 25-251; 25-301; 25-305; 25-307; 25-337; 25-341; 25-365; 25-561; 25-57...</td>
<td>Master Document and Drawing List D2101-0001, revision 7, dated 16 November 2023; CETA Airplane Flight Manual Supplement, D2101-0017, revision 5; CETA Maintenance Manual Supplement, D2101-0016, revision 5, dated 16 November 2023; later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>RNP Approaches are only certified down to LNAV minima. RNP Approaches down to LNAV/ VNAV or LPV are forbidden. Limited to SN 550-6156-Aertyp (EASA 21J. 091 minor modification n° 00112 Ed.00, associated with Service Bulletin ref. 101980); is a prerequisite for this Approval. Prior to installation of...</td>
<td>19/04/2024</td>
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<td>10084203</td>
<td>SKYTEC AEROSPACE GmbH</td>
<td>0</td>
<td>737-800</td>
<td>186Y Passenger Seating Layout</td>
<td>This modification introduces agarresting a 186-passenger seating configuration.</td>
<td>The change is defined with revision A of Skytec Master Document ListDocument MDL-B738C111-0003Maintenance on the associated final noise/ or emissions levels of the product are unchanged remain applicable to this certificate/ approval.</td>
<td>The change is defined with revision A of Skytec Master Document ListDocument MDL-B738C111-0003Maintenance on the associated final noise/ or emissions levels of the product are unchanged remain applicable to this certificate/ approval.</td>
<td>16/04/2024</td>
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<td>10084306</td>
<td>KNS AVIATION ENGINEERING EUROPE Li</td>
<td>REV. 1</td>
<td>A320-214</td>
<td>Equipment &amp; Furnishings - Cabin Reconfiguration (180 Y/C)Installation of ACRO series 9 recline economy class seats. Cabine-configuration of A320 aircraft to outfit ACRO series 9 recline economy seats for 180 passengers.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: JAR 25 Change 13 for 25.562(j)(c)(i) – JAR 25.562 (c)(s)(i) are excluded, as these subject aircraft is.</td>
<td>Master Document List, Equipment &amp; Furnishings - Cabin Reconfiguration, 23E156-MDL-001, Revision 02, dated 17th April 2024 (=) for later revisions of the above listed document(s) approved/ accepted.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/04/2024</td>
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<td>10084306</td>
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<td>10084313</td>
<td>GFD GmbH</td>
<td></td>
<td>35A, 36A</td>
<td>CMDL FWD Antenna Provision (Variant)</td>
<td>Installed in accordance with Engineering Order, 23-072-EO-K1, Rev. 01, 12.02.2024Master Data List, 23-072-MDL-01, Rev.0.1, 15.02.2024</td>
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<tr>
<td>10084325</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td>ATR 72-212, ATR 72-212 A, ATR 72-202, ATR 72-201, ATR 43-600, ATR 73-101, ATR 42-320, ATR 42-400, ATR 42-200, ATR 42-300</td>
<td>Mode 5 F/E/SHS &amp; ADS-B Upgrade with MOD A or MOD B configurations.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25 Amendment 28paragraph 4.1.25.26 25.251.24.251.301.20.303.25.305.25.307.25.337.25.341.25.365.25.66.</td>
<td>Engineering Order, Ref. 3000171-EO, Is. 1, Date: 22.04.2024Aircraft Certification Documents, Ref. 3000171-DCC, Is. 1, Date: 22.04.2024Aircraft Supplements, Ref. 3000171-AM-A, Is. 2, Date: 12.04.2024 and Ref. 3000171-AM-B, Is. 1, Date: 16.04.2024Aircraft Documents, Ref. 3000171-CA-A, Is. 1, Date: 15.12.2023 and Ref...</td>
<td>11. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/04/2024</td>
<td>Active</td>
<td>10084325</td>
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<tr>
<td>10084330</td>
<td>AIRBUS S.A.S.</td>
<td></td>
<td>A330-243</td>
<td>Airbus STC Solutions MOD A0278This modification describes the preparation of the seat configuration, as prerequisite:</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25 Amdt 4paragraph 1.1, 29.301.303, 305.307.321.337.341.365.471, 561, 562.571.601, 603.605, 6...</td>
<td>Certification Summary Report reference: SRSA-23017 Issue F dated 10 April 2024Flight Manual for EASA installations for A330-243Aircraft Approval Sheet MAS n° A0278 issue 1.Airplane Flight Manual supplement for MSN 0168..</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/04/2024</td>
<td>Active</td>
<td>10084330</td>
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<tr>
<td>10084331</td>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td>A320-251N</td>
<td>186 PAX seats with ISPSS</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25 Amdt 4paragraph 1.1, 29.301.303, 305.307.321.337.341.365.471, 561, 562.571.601, 603.605, 6...</td>
<td>Instructions for Continued Airworthiness (CA) N° 41198-01-A-ICA Flight Manual for EASA installations for A320-251N prior to installation.</td>
<td>The embodiment of this change/repair requires the following AV configuration as pre-requisite: Installation of SABENA TECHNICS BGC Flight Manual for EASA installations for A320-251N prior to installation.</td>
<td>24/04/2024</td>
<td>Active</td>
<td>10084331</td>
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</tbody>
</table>
List of EASA Supplemental Type Certificates

easa.europa.eu

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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC no.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
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<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
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<tbody>
<tr>
<td>10084366</td>
<td>BLUE SKY ADVENTURES</td>
<td></td>
<td></td>
<td></td>
<td>Installation of transparent sliding door</td>
<td>The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements:- Special Condition(s): D-50/60535/15 Installation of Overhead Seats - Equivalent Safety Finding(s): D-90-40050/51 Obstruction of Type II Emergency Exit - The CB...</td>
<td>Flight Manual Supplement Ref. AD/C-91-001-APM A 04/04/2024 Maintenance Manual Supplement Ref. AD/C-91-001-AMM A 23.3.2024 Installation Instructions Ref. AD/C-91- E2/004A 18.03.2024 Master Document List Ref. AD/C-91-001-MDL A 04/04/2024.</td>
<td>24/04/2024</td>
<td>Active</td>
<td>10084366</td>
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<tr>
<td>10084369</td>
<td>ROSEN SUNVISOR SYSTEMS L.L.C.</td>
<td>FAA STC</td>
<td>SAD137NM</td>
<td>SEE FAA AML</td>
<td>Installation of sun visor</td>
<td>The Certification Basis for the original product remains applicable to the certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval w...</td>
<td>ROSEN Product Development, Inc. Drawing List No. RCS-300-A00-00 DL, Revision 8, dated 14 February 2005&lt;&gt;(&gt;)&lt;1 yr later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU/ USA Bilateral Agreement.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/04/2024</td>
<td>Active</td>
<td>10084369</td>
</tr>
<tr>
<td>10084379</td>
<td>JET AVIATION AG</td>
<td>REV. 1</td>
<td>737-8</td>
<td></td>
<td>VVIP Cabin Interior Completion The modification installs a full VIP interior in green Boeing 737-aircraft. The aircraft is completed to accommodate a maximum of 52 passengers and 4 Cabin Crew. Rev. 1 corrects typo.</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- Special Condition(s): D-50/60535/15 Installation of Overhead Seats - Equivalent Safety Finding(s): D-90-40050/51 Obstruction of Type II Emergency Exit - The CB...</td>
<td>- Master Data List JBCS-M-03768-M01 ISSUE E - Aircraft Flight Manual Supplement JBCS-M-03768-PM-01 ISSUE C - Instructions for Continued Airworthiness JBCS-M-03768-CA-01 ISSUE A - Cabin Crew Data JBCS-M-03768-CD-01 ISSUE C - Master Minimum Equipment List JBCS-M-03768-EM-01 ISSUE C Cor later rev.</td>
<td>Private Operation (Not-for-Hire, Non-Commercial)/(Non-Smoking/Non-Max occupancy: 32 Pax, 4 Flight crew, 4 Cab Crew) Requirements - Exterior Exit Marking - an approved exterior paint finish matching the exit configuration of the cabin considering the deactivated exits must be applied. Changes to the...</td>
<td>16/05/2024</td>
<td>Active</td>
<td>10084379</td>
</tr>
<tr>
<td>10084383</td>
<td>COMPANIA ESPANOLA DE TECNOAERONA UTICA, S.A.</td>
<td>EC135 P2, EC135 P2+</td>
<td></td>
<td></td>
<td>Installation of a Video Transmission System in accordance with Master Document and Drawing List D2306-0001</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: CS-27 Amendment 2 to 27.1.7; 2.7.2; 2.7.21; 2.7.25; 27.29; 27.45; 27.141; 27.251; 27.301;27.303; 27.302...</td>
<td>- Rotorcraft Flight Manual Supplement D2306-0014 Edition 03 - Master Document and Drawing List D2306-0017 Edition 03 - Maintenance Manual Supplement D2306-0013 Edition 03 or later revisions of the above listed document(s) approved/accepted on behalf of EASA</td>
<td>The usage of the Video Transmission System is limited to VFR operations prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness...</td>
<td>29/04/2024</td>
<td>Active</td>
<td>10084383</td>
</tr>
<tr>
<td>10084407</td>
<td>KITCHENER AERD AVIONICS LIMITED/da Vinci MOD CANADA MOD CENTER</td>
<td>TCCA-LSA23-08BD</td>
<td>560XL</td>
<td>Export Configuration - CTV and SSDTU as per TCCA STC D-LSA23-08BD upgrade of an existing CTV to a L3 Harris SRI/TVR 25 SSCR with Internal Recorder Independent Power Supply (IRPS) - replacement of an existing Universal Avionics Data Transfer Unit (DTU) with a Solid State Data Transfer Unit (SSDTU).</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:- The following paragraph(s) at a later amendment: CS 25.1447(C) 4AAMt 28. The requirements for environmental protection and the...</td>
<td>Avionics Design Services Ltd. Master Drawing List MC-20343 Revisions for CAB&amp;-Canada MD Center Instructions for Continued Airworthiness/CA 22043-1 Revisions No. dated July 10, 2023 or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Imply.</td>
<td>Applicable to Textron Aviation 560XL S/N 5605101 Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/06/2024</td>
<td>Active</td>
<td>10084407</td>
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</tbody>
</table>
The Type and OSD Certification Bases (CB) for the original product as amended by the following additional or alternative airworthiness requirements-The following paragraph(s) at a later amendment: CS 25.582(c)(5) and CS 25.582(c)(6) at amendment 11. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.

Installation of EFIS - Gammm 700 TXi suite on L410 aircraft.

Seat Bottom Cushion Replacement for CS 25.562(a), (i)(2) and CS 25.785(a) compliance.

Modification of the FVFD area (DR1-DR2) passenger seats layout currently installed in B/C configuration to obtain a 347 Y/C passenger seats configuration currently installed. The configuration (Report n° 139A0050CP050/03) Dual Host Leonardo Helicopters kit p/n 4G2591F0011. Directional Finder Antenna installed within Leonardo Helicopters kit p/n G3450T05011. The SX-6 with Filr Camera Leonardo Hc...
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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>10084453</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING GES ING. WALTER STARZACHER Ges</td>
<td>737-8</td>
<td></td>
<td>Galaxy 7 installation: this modification replaces installed Galaxy 7 units with new AMES Galaxy 7 units on Boeing 737-8aircraft.</td>
<td>The Type and OSD Certification Bases (CB) for the original product does not apply to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval:..</td>
<td></td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the Row 44 Wi-Fi antenna installation and associated fuselage reinforcement members installed by the Row 44 modification (EASA STC 10033185, FAA STC ST2839CH) and localized skin reinforcement between STA727D+13.4 and STA727J+10 introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/05/2024</td>
<td>Active</td>
<td>10244432</td>
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<tr>
<td>10084454</td>
<td>BOEING IRELAND LIMITED</td>
<td>737-800</td>
<td></td>
<td>Removal of the Row 44 Wi-Fi antenna installation and associated fuselage reinforcement members installed by the Row 44 modification (EASA STC 10033185, FAA STC ST2839CH), and localized skin reinforcement between STA727D+13.4 and STA727J+10.</td>
<td>The Type and OSD Certification Bases (CB) for the original product do not apply to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval:..</td>
<td></td>
<td></td>
<td>A pre-requisite to this change is that the Row 44 Wi-Fi antenna installation and associated fuselage reinforcement members installed by the Row 44 modification (EASA STC 10033185, FAA STC ST2839CH) and localized skin reinforcement between STA727D+13.4 and STA727J+10 introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/05/2024</td>
<td>Active</td>
<td>10244454</td>
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<tr>
<td>10084455</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>A300F4-606ST</td>
<td>ST34MOD0053 - FMCDU CMA-9000 installation/ the STC covers replacement of the current Flight Management Computer Display Unit (FMCDU) with the new FMCDU CMA-9000.</td>
<td>The Type and OSD Certification Bases (CB) for the original product does not apply to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval:..</td>
<td>- Chance Certification Plan (CCP) ST34MOD0053 Issue 5 dated 28-APR-24.</td>
<td></td>
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<td>Prior to installation of this change/repair it must be determined that the Row 44 Wi-Fi antenna installation and associated fuselage reinforcement members installed by the Row 44 modification (EASA STC 10033185, FAA STC ST2839CH) and localized skin reinforcement between STA727D+13.4 and STA727J+10 introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/05/2024</td>
<td>Active</td>
<td>10244455</td>
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<tr>
<td>10084456</td>
<td>PROTOPLANE SPECIAL MISSION AIRCRAFT</td>
<td>P2006T</td>
<td></td>
<td>Lidar installation in a Tecnam P2006T.</td>
<td>The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval:..</td>
<td>Master Data List, Doc. No. PP-156-MDL-015 Supplement to Aircraft Flight Manual, Doc. No. PP-156-AMF-010 (installation manual), Doc. No. PP-156-IM-010 (installation manual):</td>
<td></td>
<td></td>
<td>Prior to installation of this change/repair it must be determined that the Row 44 Wi-Fi antenna installation and associated fuselage reinforcement members installed by the Row 44 modification (EASA STC 10033185, FAA STC ST2839CH) and localized skin reinforcement between STA727D+13.4 and STA727J+10 introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/05/2024</td>
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<tr>
<td>10084496</td>
<td>RMC CANADA, Inc.</td>
<td>TCCA SH25-13 ISSUE 2</td>
<td>369FF 369E 369D</td>
<td>Installation of Integrated Health and Usage Monitoring System (HUMS)</td>
<td>The Type and OSD Certification Bases for the original product remain applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/ approval:..</td>
<td>MD 369 Master Data List, RMI1-000386, Revision 6 dated 19 April 2024</td>
<td></td>
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<td>Prior to installation of this change/repair it must be determined that the Row 44 Wi-Fi antenna installation and associated fuselage reinforcement members installed by the Row 44 modification (EASA STC 10033185, FAA STC ST2839CH) and localized skin reinforcement between STA727D+13.4 and STA727J+10 introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/06/2023</td>
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<td>STC Holder</td>
<td>STC Type</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC0192NY</td>
<td>777-200</td>
<td></td>
<td>Installation of a Panasonic Avionics Corporation In-Flight Entertainment System - Configuration:TL10648-01 for exLife system installation - Configuration: 2 TL10974-01 exLife system installation and a xW system installation (wireless audio/video streaming).</td>
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<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC ST01007DE</td>
<td>GV-SP</td>
<td>Gv</td>
<td>Installation to Replace Nickel Cadmium Original Equipment Batteries with Valve Regulated Lead-Acid Batteries</td>
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<td>FREIBALLONSPO RT VEREN MÜNSTERUND MÜNSTERLAND e.V.</td>
<td>NL-STU/1000</td>
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<td>Introduction of Load Ring LO-01 Villker</td>
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<td>BALLONS CHAIZE</td>
<td>FBALOONS G</td>
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<td>Compatibility of Schroeder Fire Balloons Sky Heart with Ballon ChaizeA201 basket and Cameron MX53 burner</td>
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<td>AIRBUS S.A.S.</td>
<td>A350-941</td>
<td></td>
<td></td>
<td>MOD A0287 – Modify cabin configuration for ITA Airways Themodification project is a conversion of Airbus A350 armoured formation by ITA Airlines. It consists in changing the cabin layout between doors 2 and doors 3 added after (Cabin Configuration)</td>
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</table>

**Certification Basis**

- The Type and OSD Certification Basics for the original product are amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS25.1585 The requirements for environmental protection and the associated certified noise and/or emissions levels of the product remain applicable to this certificate/approval.
- Certification Program LO-01 Villker dated 07.05.2004 Th manual supplement LO-01 Villker dated 07.05.2004 or later revisions of the above listed document(s) approved/accredited by the EASA system. Limited to competition and training flights, no flights with passengers. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product. Prior to installation of this change/repair it must be determined that the relationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the airworthiness of the product.
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<td>AIR PROJET</td>
<td>S-1S, S-2A, S-2B, S-2C, S-2S</td>
<td>BERINGER PITTSPAPPRO20040 upgrades the original wheel and brake system for all Pitts aircraft models with BERINGER wheel and brake system.</td>
<td>10084525</td>
<td>The Type and OSD Certification Bases (CB) for the original product.</td>
<td>Installed in accordance with document number: APFC-0040, Installation and control manual, Revision 1, Dated 2nd May 2024.</td>
<td>None</td>
<td>17/05/2024</td>
<td>Active</td>
<td>10184620</td>
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<tr>
<td>JADE ENGINEERING PTE LTD</td>
<td>REV. 1</td>
<td>Three Class Interior Modification This design change includes an BBC18PEY/13867 interior modification with ETISO C127 seats from Collins Aerospace, onboard fourteen (14) Airbus A320-251N aircraft.</td>
<td>10084538</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>Compliance Document List, JD-348C03-1 Revision B, 29th May 2024.</td>
<td>IN-SEAT POWER SUPPLY SYSTEM (ISPSS): The ISPSS for PEDs carried by the passengers must be turned off during take-off and landing.</td>
<td>28/05/2024</td>
<td>Active</td>
<td>10204530</td>
<td></td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>A330-223</td>
<td>A330-223 cabin reconfiguration from 312 Y/C to 345 Y/C seats.</td>
<td>10084550</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>Master data list, Ref. 41025-01-A-MDL-F01 issue R00 dated 16 May 2024.</td>
<td>Non-electrical floor path marking system (EASA STC 10014507) shall be installed. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the aircraft...</td>
<td>24/05/2024</td>
<td>Active</td>
<td>10204550</td>
<td></td>
</tr>
<tr>
<td>HELIKOPTER AIR TRANSPORT GESELLSCHAP Am b. H.</td>
<td>EC135 T3</td>
<td>Installation of Jensstab Avionics EB1 Multipurpose Airborne Computer (IMPC), external LIT404 antenna and internal Wireless Local Area Network (WLAN) antenna in accordance with Engineering Order 34-006</td>
<td>10084555</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>- Aircraft Flight Manual Supplement AFMS 34-006 Revision 0-Engineering Order EO 34-006 Revision 0: Instructions for Continued Airworthiness ICA 34-006 Revision 0 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the aircraft...</td>
<td>24/05/2024</td>
<td>Active</td>
<td>10204550</td>
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<tr>
<td>SABENA TECHNICS BGC</td>
<td>A330-343</td>
<td>Cabin Reconfiguration to 391 Pax (68/C, 385 Y/C).</td>
<td>10084563</td>
<td>The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:</td>
<td>Master Data List 41029-01-A-MLD R00 for Continued Airworthiness ICA 34-006 for later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of the change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/or repair will introduce no adverse effect upon the aircraft...</td>
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<td>Airbus Transport International SNC</td>
<td>Active</td>
<td>- Change Approval Sheet (CAS) ST25MOD0026E58 Issue 1 dated 23 May 2024. Change Certification Plan (CCP) ST25MOD0026E58 Issue 3 dated 15 May 2024. Approved Document Dossier (ADD) ST25MOD0026E58. ADD Issue 3 including AFM, WBM and MME. supplements as enclosed in independent Check Sheet 2302 Issue 2 sign...</td>
<td>21/05/2024</td>
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<td>Havaclik Muhendislik ve Sertifikasihizm Etleri Sanayi ve Ticaret Anonimsirketi</td>
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<td>Master Data List, MDL-751190 Issue IR, 30th March 2024, later revisions of the above listed document(s) approved/accepted under EASA system.</td>
<td>29/05/2024</td>
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<td>AMAC Aerospace Switzerland AG</td>
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<td>Master Data List, MDL-10074237 Issue 1 dated 15 May 2024. Change Certification Sheet 2302 Issue 2 sign...</td>
<td>30/05/2024</td>
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<td>Delta Engineering Corporation</td>
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<td>Master Data List, MDL-10074237 Issue 1 dated 15 May 2024. Change Certification Sheet 2302 Issue 2 sign...</td>
<td>31/05/2024</td>
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<td>Dassault Aviation Business Services S.A.</td>
<td>Active</td>
<td>Limited to aircraft serial number 60. This STC constitutes an Alternative Method of Compliance (AMOC) to the action required by EASA AD 2014-0052, sect. “Required Action(s) and Compliance Time (“x”) (paragraph 1) Prior to installation of the change/repair it must be determined that...</td>
<td>03/06/2024</td>
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<tr>
<td>DOA AESGLOBAL LIMITED</td>
<td>- The Change is defined with Issue 4 of AESGLOBAL Master DataListAES-0385-07-MDL. Maintenance shall be accomplished in accordance with Issue 0 ofAESGLOBAL Instructions for Continuous Airworthiness DocumentAES-0385-07-01 or later revisions of the above listed document(s) approved/accepted under the EASA, Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/06/2024</td>
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<td>AMAC</td>
<td>- The Change is defined with Revision B of AMAC Master DataList Document MDL-EO09-2500 - The aircraft shall be maintained in accordance with Revision A of AMACInstructions for Continued Airworthiness document ICA EO09-2500 - The aircraft shall be operated in accordance with Revision B of AMACAirplane...</td>
<td>04/06/2024</td>
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<td>STERLING CORPORATION/ VIA STERLING HELICOPTER</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/ or repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/06/2024</td>
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<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>Flight limitations associated with the Portable EFB Mount are detailed in the Flight Manual Supplement FMS-1836-001. Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/06/2024</td>
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<tr>
<td>SAFRAN CABIN INC</td>
<td>Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.Embraer Document Number STD2023-190-01818 must be inst...</td>
<td>10/06/2024</td>
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**List of EASA Supplemental Type Certificates**

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<tr>
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>DOA AESGLOBAL LIMITED</td>
<td>24BC / 250YC Dual Class Passenger Seating Layout</td>
<td>The Type and OSE Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS 25.1, 25.29, 25.303, 25.395 (a) – (c), 25.307 (a), 25.321, 25.337, 25.369 (a) – (g), 25.561, 25.601, 25.603, 25.604, 25.657, 25.661, 25.662, 25.109, 25.111, 25.115</td>
<td>- The Change is defined with Issue 4 of AESGLOBAL Master DataListAES-0385-07-MDL. Maintenance shall be accomplished in accordance with Issue 0 ofAESGLOBAL Instructions for Continuous Airworthiness DocumentAES-0385-07-01 or later revisions of the above listed document(s) approved/accepted under the EASA, Prior to installation of this change/repair it must be determined the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/06/2024</td>
<td>Active</td>
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<tr>
<td>AMAC</td>
<td>Cabin Reconfiguration</td>
<td>The Type and OSE Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements - The following paragraph(s) at a later amendment: CS 25.853, CS 25Appendix B 25.10, 25.26, 25.30, 25.36, 25.50 and 25.60.19, Amendment 19: The related...</td>
<td>- The Change is defined with Revision B of AMAC Master DataList Document MDL-EO09-2500 - The aircraft shall be maintained in accordance with Revision A of AMACInstructions for Continued Airworthiness document ICA EO09-2500 - The aircraft shall be operated in accordance with Revision B of AMACAirplane...</td>
<td>04/06/2024</td>
<td>Active</td>
<td>10084623</td>
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</tr>
<tr>
<td>STERLING CORPORATION/ VIA STERLING HELICOPTER</td>
<td>Installation of replacement LED recognition lights, LED wingtip lights, LED beacon lights, LED taxi lights, and LED landing lights</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Sterling Helicopter Master Drawing List (MDL), Document No.ST-0791-02, Rev., dated 23 January 2024&lt; (...&gt; Maintenance to be performed in accordance with Sterling HelicopterInstructions for Continued Airworthiness (ICA) Document Number ST-0791-03, Rev. &gt;, dated 5 February 2024&lt; &gt;, or later revision...</td>
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<td>Active</td>
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<tr>
<td>EIRTECH AVIATION SERVICES LIMITED</td>
<td>Portable EFB Mount on 737 Series</td>
<td>The Type and OSE Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements - Special Condition(s): SC-25.1585-01: Mitigation of flight deck fixwiring, signalling from lithium batteries that are not part of the aircraft system - The follow...</td>
<td>Master Document List – MDL/1836-001 Issue 2, dated 6th June 2024Flight Manual Supplement FMS-1836-001 Issue 2, dated 14 May 2024&lt; later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
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<tr>
<td>SAFRAN CABIN INC</td>
<td>Interim Installation and Modifications</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval.</td>
<td>Index List No. K151, Rev. A, dated 15 May 2024&lt; (...&gt;Instructions for Continued Airworthiness (ICA), Report No. CDRK151-15, Rev. NC&lt;(...&gt; dated 28 March 2024&lt; or later revisions of the above listed document(s) approved/accepted under the EASA in accordance with the Technical ImplementationPr...</td>
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<td>Description</td>
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<td>Limitations/Conditions</td>
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| GULFSTREAM AEROSPACE CORPORATION| FAA STC | Installation of an Aircraft ATG 4000 High Speed Internet                  | Certification Basis for the original product remains applicable to this certificate/approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the product are unchanged and remain applicable to this certificate/approval. | Gulfstream Aerospace Corporation Index List ECE130308/16, Revisions E-dated 22 April 2024≤(>,<)≤(>,<)≥Airplane Flight Manual Supplement ECE13030/4, Rev.A, dated 18 January 2023≤(>,<)≥Instructions for Continued Airworthiness. Document ECE1303A06, Revision A, dated 31 July 2012≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>,<)≥1≤(>，“ practically or universally applicable or used, etc.“}
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<th>Certification basis</th>
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<th>Limitations/Conditions</th>
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<td>AVINCS AVIATION ITALIA S.P.A</td>
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<td>CL-215-6811 (CL-415 VARIANT)</td>
<td>Change No 34-50-2307-A: CL-415 Flight Management System Installation/Installation of a Flight Management System (FMS) with PBN capabilities</td>
<td>The Type and OSI Certification Bases for the original product are amended by the following additional or alternative airworthiness requirements: The following paragraph(s) at a later amendment: CS-ACNOS Issue 4</td>
<td>Flight Manual Supplement FMS-34-50-2307-A Rev. 00</td>
<td>Instructions for Continued Airworthiness ICA-34-50-2307-A Rev. 00</td>
<td>Certification Programme CTP-34-50-2307-A Rev. 04</td>
<td>Master Document ListML-34-50-2307-A Rev. 000</td>
<td>Later revisions of the above listed document(s) approved</td>
<td>23/04/2004</td>
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<td>2003-428</td>
<td>GOODRICH CORPORATION</td>
<td>FAA ST00933SE:D</td>
<td>737-600/-700/-800/-900</td>
<td>Cockpit door surveillance system</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate approval. The requirements for environmental protection and the associated certified noise and/or emission levels of the product are unchanged and applicable to this certificate approval without any imp..</td>
<td>- Master Data List, D2003-073, Revision B, dated 22 October 2003. AFMS, D2003-148, Revision New, dated 4 September 2002. Instructions for Continued Airworthiness (ICA), D2003-149, Revision New, dated 17 September 2003 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change/repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>2004-4159</td>
<td>ANORA INTERNATIONAL LLC</td>
<td>FAA STC ST01536A</td>
<td>737-300</td>
<td>Installation of Ancra’s Cargo Handling System on Channel Express B737-300SF aircraft</td>
<td>The modification comprises a Sensenich 650XCS12, or M650XCS12 propeller having a diameter in the range 67 to 69 inches, and pitch between 48 and 52 inches. The modification kit includes spinner assembly C-2326, and documents/technical/installation instructions and, drawing no. A-1076.</td>
<td>AMOCA Drawing No. DL83734, dated 19 August 2003.</td>
<td>None.</td>
<td>23/04/2004</td>
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<td>AAN NO. 26826</td>
<td>SENSENICH PROPELLER MANUFACTURING COMPANY, Inc.</td>
<td>FAA STC SA161ME</td>
<td>F150L A150M A150L</td>
<td>The modification comprises a Sensenich 650XCS12, or M650XCS12 propeller having a diameter in the range 67 to 69 inches, and pitch between 48 and 52 inches. The modification kit includes spinner assembly C-2326, and documents/technical/installation instructions and, drawing no. A-1076.</td>
<td>- Master Data List, D2003-073, Revision B, dated 22 October 2003. AFMS, D2003-148, Revision New, dated 4 September 2002. Instructions for Continued Airworthiness (ICA), D2003-149, Revision New, dated 17 September 2003 or later revisions of the above listed document(s) approved/accepted under the EASA system.</td>
<td>None.</td>
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<td>EASA A.S.00175</td>
<td>DASSAULT FALCON SERVICE</td>
<td>FALCON 20 EQUIPPED WITH COLLIN</td>
<td>Installation of modified cockpit controls, kits and panels; see DFS/Change Approval Sheet2385 Issue A dated 15 July 05</td>
<td>-see Change approval sheet DFS 2385 Issue AMM Supplement DF SIDE692 F2 - Part 76-10-00 and associated list of affected technical documents</td>
<td>- This STC is applicable to Falcon 20 equipped with Collins Proline IV/Avionics only. This STC is approved only for the configuration as defined in ModSheet DFS 2385 Issue A3. This STC is approved only for the product configuration as defined in the associated data sheet referenced to the paragraph...</td>
<td>17/06/2005</td>
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<td>EASA A.S.00182</td>
<td>DASSAULT FALCON SERVICE</td>
<td>FALCON 20 EQUIPPED WITH COLLIN</td>
<td>Deviation to STC EASA IM A.S.00237 (Rockwell Collins Proline IV/DF3Installation) iaw Change Approval Sheet DFS 2524 Issue A dated 15 July 05</td>
<td>-see Change approval sheet DFS 2524 Issue AMM chapter 5 DFS/SE/5F2 A and associated list of affected technical documents</td>
<td>- This STC is applicable to Falcon 20 equipped with Collins Proline IV/Avionics only. This STC is approved only for the configuration as defined in ModSheet DFS 2524 Issue A3. This STC is approved only for the product configuration as defined in the associated data sheet referenced to the paragraph...</td>
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<td>EASA A.S.00186</td>
<td>CRANFIELD AEROSPACE SOLUTIONS LIMIT</td>
<td>CAA UK RESPONSIBILITY</td>
<td>JETSTREAM 4100</td>
<td>Enabling of Traffic Collision Avoidance System II (TCAS II) in accordance with Cranfield Aerospace Ltd. Document CA/DDR2122, issue B dated June 2004 or later EASA approved revision. Flight Manual Supplement Cranfield Supplement No 1 to CAA approved Flight Manual Document No. J41-01.</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E: irrevocable and associated data shall remain valid until superseded, withdrawn or otherwise terminated. 2. If transfer of this certificate is requested under Part 21A, it will be replaced. 3. This STC is approved only for the product...</td>
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<td>Description</td>
<td>Certification basis</td>
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<td>Limitations/Conditions</td>
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<td>EASA.A.S.00196</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
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<td>A319-112</td>
<td>Cabin reconfiguration into an executive 44 pax, in accordance with the Modification Approval Sheet (MAS) Ref. SOG-00-830, issue 4, dated September 20, 2004</td>
<td>The Document Ref.: DIT-N II 2004/1728/OS Revision B must be incorporated as a supplement to the Airplane Flight Manual (AFM)</td>
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<td>EASA.A.S.00198</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 1</td>
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<td>BOEING 747-400</td>
<td>Air France B747-400 „ULYSSE“ cabin reconfiguration. This is a revision to STC Ref EASA.A.S.00198, to extend the applicability to the aircraft B747-400 F-GIT H, F-GIT L, F-GIT M, and F-GEA S. 24164. The general description is included in the Air France Letter BL.110 002/EB and in the AEO...</td>
<td>1. The approval of this modification is limited to the Boeing B747-400 N80 24969, 24990, 25344, 25600, 25601, 25602, 25606, 25609, 25611, 25614, 25154. The limitations and conditions defined in the EASA Datasheet are applicable. 2. A copy of this STC and of the CAS...</td>
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<td>EASA.A.S.00216</td>
<td>AERODATA AG</td>
<td>EASA A.S. 00216</td>
<td>560, 550, 560</td>
<td>Installation of an Aerophone Iridium Satcom System in accordance with Master Drawing List Doc. No.: MDL- AVC-082 and Drawing List Doc. No.CLS-A C-09, Rev. 0 (or later EASA approved revision).</td>
<td>1. Operations Instruction: Aerodata Flight Manual Supplement AFM-058, Rev. 02, Instruction for Continued Airworthiness Doc. No. HB-AC-708, Rev. 023. Installation Instruction: Install and Testing Instruction Doc. No. REP-AC-1018, Rev. 014. This STC is approved only for the product configuration as defined...</td>
<td>1. Operation Instruction: Aerodata Flight Manual Supplement AFM-071, Rev. 02, Instruction for Continuous Airworthiness Doc. No. AMM-758, Rev. 03. Installation Instruction: Modification Order Doc. No. HB-AC-7564, This STC is approved only for the product configuration as defined in the approved design...</td>
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<td>EASA.A.S.00217</td>
<td>AERODATA AG</td>
<td>560</td>
<td>Installation of a TCAS 2000 in accordance with Master Drawing List Doc. No. MDL-VC-081 (or later EASA approved revision).</td>
<td>1. Operation Instruction: Aerodata Flight Manual Supplement AFM-071 Rev 02 Instruction for Continuous Airworthiness Doc. No. AMM-758, Rev. 03 - Installation Instruction: Modification Order Doc. No. HB-AC-7564, This STC is approved only for the product configuration as defined in the approved design...</td>
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<td>EASA.A.S.00220</td>
<td>MOTORFLUG BADEN BADEN GmbH</td>
<td>F27 M500</td>
<td>Installation of Trimble LNL 2101 IQ Approach Plus GPS in accordance with E/O No. 114112 of the Fa. Aero Technic</td>
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### List of EASA Supplemental Type Certificates

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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>BEECH 400A</td>
<td>Installation of a ELT on A330. All documents related to this STC are listed in the Master Drawing List No. 00848-JAA-MDL Rev. B dated 23/09/2004 or later EASA approved revision.</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>BEECHJET 400A</td>
<td>Installation of an Artex ELT 406 in accordance with Engineering Order Doc.: 2017016-E0-01. Issue 2 or later EASA approved revision.</td>
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<td>Installation of an Artex ELT 406 in accordance with Engineering Order Doc.: 2017016-E0-13. Issue 2 or later EASA approved revision.</td>
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<td>EASA.A.S.00243</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 2</td>
<td>A340-600</td>
<td>Installation of a Patient Transport Compartment (PTC) having as Certification Basis JAR-25 Change 15 as defined in JAA TCDS/92/0004 Issue 33</td>
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<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.A.S.00265</td>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>REV. 1</td>
<td>BOEING 737-500, BOEING 737-300, 737-400</td>
<td>Installation of Honeywell EGPWS Mark V</td>
<td>EAD Master Drawing List ref. 00772-JAA-A-ADL revision GEAD AFM Supplement ref. DC113-0772-00 revision HEAD-MPO Supplement ref. DC114-0772-00 revision Ac1 later EASA approved revisions of those documents</td>
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<td>REV. 2</td>
<td>B300</td>
<td>Flight Data Recorder (FDR) integration into Hawker Beechcraft King Air B300</td>
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<td>10013933</td>
<td>CESSNA 525</td>
<td>Certification basis</td>
<td>Installation of an elementary SCANDINAVIAN</td>
<td>Date of Aircraft modification for 8.33 kHz spacing</td>
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<td>REV. 1</td>
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<td>Description</td>
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<td>EAD AEROSPACE</td>
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<td>737</td>
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<td>Installation of an elementary surveillance / enhanced surveillance ATCtransponder system. This STC supersedes the STC EASA.A.S.00339 Revision2</td>
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<td>SOGEMA SERVICES</td>
<td>REV. 1</td>
<td>AIRBUS A310-304</td>
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<td>Aircraft modification for 8.33 kHz spacing on VHF system in accordance with Modification Approval Sheet SOG 23-878</td>
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<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td>A330-203</td>
<td>VIP Cabin Installation</td>
<td>All documents related to this STC are listed in the Master Data List Drawing No MDL-01-502 Issue A (or later EASA-approved revision).</td>
<td>1. The approval of this modification is limited to the Airbus A330-203 MSN 487 which is used as a private aircraft. The limitations and conditions defined in the EASA Datasheets are applicable. A copy of this STC and of the MEL, AFM must be enclosed in the flight the modified aircraft. Any alteration of the STC definition must be m...</td>
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<td>EASA.A.S.00348</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td>A330-203</td>
<td>Installation of a SATCOM live TV system.</td>
<td>All documents related to this STC are listed in the Master Data List Drawing No. MDL-01-580 Issue A (or later EASA-approved revision).</td>
<td>1. The approval of this modification is limited to the AIRBUS A330-203 MSN 487. The limitations and conditions defined in the EASA Datasheet are applicable. A copy of this STC and of the MEL, AFM must be enclosed in the flight the modified aircraft. Any alteration of the STC definition must be m...</td>
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<td>EASA.A.S.00352</td>
<td>SEFA</td>
<td>REV. 1</td>
<td>BEECH B200</td>
<td>Installation of calibration workbench-ref. STNA &quot;Margouillat1&quot; according to the 02-1104-DIR document. - Installation of calibration workbench-ref. STNA &quot;Margouillat1&quot; Revision 1: Extension of STC EASA.A.S.00352 with a minor change to another aircraft (F-GJFE s/n BB1359).</td>
<td>- Aircraft Flight Manual Supplement ref 03-1104-DIR dated November 2002 or later approved revisions.- 02-1104-DIR/REV 1</td>
<td>1. Applicable only for F-GJFA s/n BB-1370, F-GJFC s/n BB-1347 and F-GJFE s/n BB1359 aircraft. 2. Prior to installation of this modification the installer must determine that the dependencies between this modification and any other previously installed modification will introduce no adverse...</td>
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<td>EASA.A.S.00357</td>
<td>AEROS DESIGN LTD</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>172</td>
<td>Introduction of Dual Image Capture System: AED Design Modification Certificate 25-79-419 Issue 1 dated October 12, 2004 under EASA-approved revision</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, the STC and associated data shall remain valid until suspended, withdrawn or otherwise terminated. 2. If transfer of this Certificate is requested under Part 21A, 118, the STC will be released. 3. This STC is approved only for the product configuration defined in the EASA Datasheet. A copy of this STC and of the MDL, AFM must be enclosed in the file of the modified aircraft.</td>
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<td>EASA.A.S.00375</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 1</td>
<td>AIRBUS A319-115</td>
<td>A319 CJ CASTA II Cabin Reconfiguration</td>
<td>Change Approval Sheet CAS-A320-04-002 rev 01 dated February 1, 2005/AR for Industries Airplane Flight Manual Supplement (AFM-01) ref: 1. Government of France F-RBFA/ F-RBFP, Revision 02, dated February 1, 2005, or later EASA-approved revision</td>
<td>1. This STC is limited to the A319-115 CJ MSN’s 1485 and 15562. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.A.S.00376</td>
<td>EADS SOGERMA</td>
<td>REV 1</td>
<td>AIRBUS A310-304</td>
<td>Royal Jordanian ATC Mode S Enhanced Surveillance installation according to EADS Socgerma Modification Approval Sheet SOC-34-974 issue 3 dated 29 November 2004. Note: This revision reflects the transfer of this STC from EADS Socgerma Services to Socgerma Services.</td>
<td>1. The Aircraft Flight Manual Supplement No AFM-34-974 issued in 2004, or later EASA-approved revision, is a required part of this STC. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other p...</td>
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<td>EASA.A.S.00377</td>
<td>EADS SOGERMA</td>
<td>REV 1</td>
<td>AIRBUS A310-325</td>
<td>Cabin Modification from 18FC/18VC to 18BC/18VC according to the modification approval sheet MAS No S0G-25-977 Issue 1 dated October 6 2004 or later EASA-approved revision. Note: This revision reflects the transfer of this STC from EADS Socgerma Services to Socgerma Services.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>ELECTRONIQUE MAINTENANCE AERONAUTIQUE</td>
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<td>F.A.R. No51/65MA</td>
<td>Installation of VHF 8.33 kHz/572/576 MHz/ GPS/STORMSPECFOR VHF/ GPS/STORMSPECFOR UHF on Airbus A330-300.</td>
<td>- Installation of VHF 8.33 kHz/572/576 MHz/ GPS/STORMSPECFOR VHF/ GPS/STORMSPECFOR UHF on Airbus A330-300.</td>
<td>1. This modification is applicable only to SE 226 T(B) aircraft.</td>
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<td>VITROCOPT S.P.A.</td>
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<td>500, 550 SERIES</td>
<td>Installation of Honeywell ECPWS on Airbus A330-300.</td>
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<td>EASA.A.S.00419</td>
<td>AVIATION TRADERS LIMITED</td>
<td>REV. 1</td>
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<td>Modification ATL/CPT/ATM33 Issue 1 dated 29/11/2004, Business Class Double Seat p/n C2C20T555(196)48N</td>
<td>ATL, DOP-603 Issue 2 dated 29/11/2004 or later EASA approved revision</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, the STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. If transfer of this Certificate is requested under Part 21A.116, the STC will be reissued. This modification does not include approx...</td>
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<td>EASA.A.S.00426</td>
<td>MADELEC AERO S.A.S.</td>
<td>REV 3</td>
<td>AIRBUS A319-320-321</td>
<td>Replacement of the powered floor proximity emergency escapepath marking system by a FPEEPM photoluminescent system</td>
<td>Certification Document List (CDL) ref: 9908, Issue of February 20, 2006Airplane Flight Manual Supplement, ref. FMS-9908, Issue 4, dated February 27, 2006Master Minimum Equipment List Supplement, ref. M96-9908, Issue 3, dated February 27, 2006</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>LUFTHANSA TECHINIC AG</td>
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<td>A340-600</td>
<td>Installation of a system for measurement of atmospheric components</td>
<td>Certification basis as defined in Type Certification Data Sheet A2A/25/02-0004, Issue 30, Supplemental Type Definition Document STDG-03/11, Issue 01, Rev. 03, November 2004, Supplemental Maintenance Requirements as referenced in document &quot;Instructions for Continued Airworthiness&quot; Doc. NO. CDL.503-01...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EADS SOGERMA</td>
<td>REV 1</td>
<td>AIRBUS 310-325</td>
<td>Modification of ATLAS/KSSU galleys into full ATLAS standard in accordance with Modification Approval Sheet reference 500-25-958 Issue 01 dated 14 October 2004. All documents related to this STC are listed in the Master Drawing List reference MSL-25-958 Issue 01 ref:06/10/2005</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.A.S.00438</td>
<td>SCANDINAVIAN AVIONICS DESIGN GMBH</td>
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<td>222</td>
<td>Installation of ELT type Kannad 406 AF / AF-H. Reuse or exchange antenna</td>
<td>-System Description 2011/33-DE-01 11/03/2004,Techical Order 1019313-T011 rev. 8, 09/02/2003 ,Drawing List 2011/33-01-01 1/03/2004 ,issue 3, 11/03/2004 later EASA approved revision</td>
<td>1. This STC assumes that an existing antenna can be reused, or that an exchange of the antenna may be done without exchanging the doubler. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; compatibility with othe...</td>
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<td>EASA.A.S.00439</td>
<td>E.I.S.A.</td>
<td>REV 1</td>
<td>CESSNA 551</td>
<td>Installation of two Rockwell / Collins TDR 94D instead of two TDR 90 according to FDM EISA-05/3M F055 rev. 4 dated November 2004 and associated documents. New transponder location.</td>
<td></td>
<td></td>
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<td>1. Compatibility of this modification with other previously-approved modifications must be determined by the installer. 2. Applicability Cessna 551 S/N 551-0133 and S/N CESSNA 5174. 3. EASA approved Airplane Flight Manual Supplement, dated 23rd November 2004 is required. 4. STC is approved only for 1...</td>
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<td>EASA.A.S.00451</td>
<td>ATS INTERNATIONAL</td>
<td>REV 1</td>
<td>DC-9-80 (MD-80) SERIES</td>
<td>Mk VII EGPWS System Installation</td>
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<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>21/03/2006</td>
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<td>EASA.A.S.00456</td>
<td>SCANDINAVIAN AVIONICS DESIGN GMBH</td>
<td>REV. 1</td>
<td>DHC-7-110/-111 DHC-7-100/-101/-102/-103</td>
<td>Installation of VDL Mode 4 Transponder</td>
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<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked...</td>
<td>31/01/2006</td>
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<td>10053185</td>
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<td>EASA.A.S.00463</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>REV. 1</td>
<td>DORNIER 328-300 DORNIER 328-100</td>
<td>Installation of Air Ambulance Equipment, VIP interior</td>
<td></td>
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<td>1. Subject to compliance with the provisions of EC 1702/2003 PART 21 Subpart E, this Certificate and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>21/04/2005</td>
<td>Active</td>
<td>10042957</td>
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<tr>
<td>EASA.A.S.00466</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>REV. 1</td>
<td>JETSTREAM 3200 SERIES</td>
<td>INSTALLATION JETSTREAM 3100 SERIES</td>
<td>Installation of Air Ambulance Equipment</td>
<td></td>
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<td>1. Compatibility of this installation with previous installed equipment must be determined by the installer. 2. Subject to compliance with the provisions of EC 1702/2003 PART 21 Subpart E, this Certificate and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>21/04/2005</td>
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<td>EASA.A.S.00473</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
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<td>F27 MK500</td>
<td>Installation of Honeywell EGPWS Mk. VI Mk. VII</td>
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<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>05/01/2005</td>
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### List of EASA Supplemental Type Certificates

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<th>STC No</th>
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<th>Description</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tr>
<td>EASA.A.S.00495</td>
<td>F27</td>
<td>&quot;Radar System Replacement with bendix ART 2000 System&quot; No 03/03. Note: This STC supersedes EASA.A.S.00495.</td>
<td>Doc No 400-0303-000-000 Master Drawing List rev. 3 dated 10.11.2004 - AFM Supplement No. 400-0303-800-000 Rev. 3 dated 15.10.2004.</td>
<td>None</td>
<td>07/01/2009</td>
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<td>10143094</td>
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<tr>
<td>EASA.A.S.00496</td>
<td>F27</td>
<td>&quot;MK 8 GPWS replacement with MK VIII EGPSW&quot; No 04-03. Note: This STC supersedes EASA.A.S.00496.</td>
<td>Doc No 700-0403-000-000 Master Drawing list rev. 3 dated 10.11.2004 - AFM Supplement No. 700-0403-800-000 rev. 3 dated 15.10.2004.</td>
<td>None</td>
<td>07/01/2009</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.00502</td>
<td>CL-604-40163-000-S-LP</td>
<td>Installation of Honeywell EGPSW Mark V.This STC supersedes and replaces the DGAC STC No F-CM135-SF-0708.</td>
<td>ATS International Master Data List No 0124-01-2103- R13-Configuration 1 (EGPSW with internal GPS): ATS International AFM Supplement No 0124-01-A2301-R03, Configuration 2 (EGPSW with FMS): ATS International AFM Supplement No 0124-01- R2301-R02-Airworthiness Limitations as described in the ATS International.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>24/01/2005</td>
<td>Active</td>
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<td>EASA.A.S.00505</td>
<td>BOMBARDIER CL-600-2B16 (CL-601 BOMBARDIER CL-600-2A12 (CL-601 BOMBARDIER CL-600-1A11 (CL-600</td>
<td>Upgrade Dual Honeywell FMS NZ-2000- PRNAV.</td>
<td>Master Document List 100A-MD, Airplane Flight Supplement 100A- AFMS: Continued Airworthiness Plan 100C- CAWP.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>12/01/2005</td>
<td>Active</td>
<td>1013895</td>
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<td>EASA.A.S.00506</td>
<td>BOMBARDIER CL-600-2B16 (CL-601 BOMBARDIER CL-600-2A12 (CL-601 BOMBARDIER CL-600-1A11 (CL-600</td>
<td>Mode-S Transmitter Collins TDR 94D upgrade - ELBEHS: Associated Technical Documentation: Master Document List 100C-MD, Airplane Flight Manual Supplement 100C-AFMS, Continued Airworthiness Plan 100C- CAWP.</td>
<td>Master Document List 100C-MD, Airplane Flight Supplement 100C- AFMS: Continued Airworthiness Plan 100C- CAWP.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>12/01/2005</td>
<td>Active</td>
<td>1013895</td>
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<tr>
<td>EASA.A.S.00510</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>Installation of ProLine IV Integrated Avionics Suite.</td>
<td>Master Document List MDL 1001FRHAM/FO10 to DTM019, or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>23/02/2006</td>
<td>Active</td>
<td>10143096</td>
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<tr>
<td>EASA.A.S.00516</td>
<td>CL-600-2B16 (604 VARIANT)</td>
<td>Installation of an Artex ELT C406-1.</td>
<td>Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.</td>
<td>Master Document List CL604-40163-000-S-LP. Prior to installation of this change/rework it must be determined that there is no adverse impact between this change/rework and any other previously installed change and/ or repair. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>19/01/2005</td>
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<td>1013895</td>
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<td>EASA STC no.</td>
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<td>Foreign/NAA STC ref.</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.A.S.00517</td>
<td>SCANDINAVIAN AVIONICS DESIGN GMBH</td>
<td>JETSTREAM SERIES 3200 JETSTREAM SERIES 3100</td>
<td></td>
<td>Installation of Honeywell CAS 67A ACAS with dual MST 67A Mode S Transponder</td>
<td>- Engineering Order 2017230-EO-01 Issue 3; Airplane Flight Manual Doc. 2017230-AM-01 (ACAS System), Issue 2; Airplane Flight Manual Doc. 2017230-AM-02 (Mode S), Issue 2; Instruction for Continuous Airworthiness Doc.2017230-CA-01, Issue 1 or later approved revisions</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configuration shall be determined by the installer.</td>
<td>19/01/2005</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.00545</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>550</td>
<td></td>
<td>Installation of Honeywell (Bendix/King) RDR 2000 Color Weather Radar System</td>
<td>Document No. CS50-05, Operating Instruction Doc. No. CS50-05-001, dated September 29, 2004 or later EASA approved revision. Instructions for Continued Airworthiness Doc. CS50-05-0701</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configuration shall be determined by the installer.</td>
<td>27/01/2005</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.00555</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A340-300</td>
<td>Cabin refurbishment with installation of New Business-Class Seats</td>
<td>STDG 00-0417M iss. 01416L 00-0417M</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configuration shall be determined by the installer.</td>
<td>25/01/2005</td>
<td>Active</td>
<td>10070400</td>
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<tr>
<td>EASA.A.S.00577</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td>LEARJET 35A</td>
<td>Installation of Honeywell ACAS II CAS 67A and Mode-S XPDR MST 67A</td>
<td>Installation of Honeywell ACAS II CAS 67A and Mode-S XPDR MST 67A</td>
<td>Associated Technical Documentation: Engineering Order Report 2004-6, Rev. B, 03 December 2004; AFMS Report 2004-8-8 rev. A, 18 October 2004; Limitations and Conditions 1. Certification basis refer to FAA TCDS A10CE plus JAA TGL No. 9 and FAA TGL No. 132. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configuration shall be determined by the installer.</td>
<td>09/02/2005</td>
<td>Active</td>
<td>10051198</td>
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<tr>
<td>EASA.A.S.00578</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>152, A152, F152, FA152</td>
<td>Installation of Propeller MT 183 R115-2C or, Sarnenich 72CKBB-6-56 or, Sarnenich 72CKBB-6-54 in combination with silencer system LisesaR74xbr100.</td>
<td>Installation of Honeywell CAS 67A ACAS System including Dual MST 67A Mode-S Transponders</td>
<td>- Instructions for Continued Airworthiness and Installation Instructions No. E-1124-Airplane Flight Manual Supplement No. E-1123</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configuration shall be determined by the installer.</td>
<td>09/02/2005</td>
<td>Active</td>
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<td>EASA STC no.</td>
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<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.A.S.00589</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
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<td>550, 560</td>
<td>Installation of Honeywell CAS-67A ACAS II and Dual MST 67A Mode STransponder by addition</td>
<td>Document defining Modification. E.O. C550-02-0101-Airplane Flight Manual Supplement for Cessna 550/560 Bravo / 560 Series Doc. No. C550-02-601, Rev. 2, or later EASA approved revisions. Instructions for Continued Airworthiness.</td>
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<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.A.S.00599</td>
<td>SCANDINAVIAN AVIONICS DESIGN GMBH</td>
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<td>P210N</td>
<td>Installation of Stormscope WX-500</td>
<td>EASA approved Airplane Flight Manual Supplement, Doc. No. 2016188-AM-01 Issue 2 or later EASA approved issue.</td>
<td></td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.A.S.00609</td>
<td>SCANDINAVIAN AVIONICS DESIGN GMBH</td>
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<td>DORNIER 228-100/-200 SERIES</td>
<td>Installation of Honeywell EGPWS Mk VI, KMD 850 Multifunction Display and RDR2000 Weather Radar.</td>
<td>AFM Supplement 2021714-AM-01 Issue 2 or later approved revision. AFM Supplement 2021714-AM-02 Issue 3 or later approved revision. Continued Airworthiness Instructions 2021714-CA-01</td>
<td></td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<tr>
<td>EASA.A.S.0061</td>
<td>VITROCISET S.p.A.</td>
<td>REV. 1</td>
<td></td>
<td>F27</td>
<td>STC No. 02/04; Electronic flight Instrument System (EFI 550) and Vertical Gyros Installation. Note: This STC supersedes EASA.A.S.0061.</td>
<td>none</td>
<td></td>
<td>1. Instruction for Continued Airworthiness. 400-0204-400-000 Rev. IR dated 30.04.2004 400-0204-800-000 Rev. IR dated 30.04.2004 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.A.S.00610</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
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<td>SOCATA TB 20</td>
<td>Installation of Transponder Garmin GTX 330 - Mode S</td>
<td>AFM Supplement TB 25-01-0601 AFM Supplement TB 25-01-0602-Instructions for Continued Airworthiness TB 20-01-0701</td>
<td></td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<tr>
<td>EASA.A.S.00645</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
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<td>LEARJET 55</td>
<td>Installation of Honeywell ACAS II CAS 67A and Mode S XPDR MST 67A</td>
<td>Engineering Order Report 2004-5, Rev B, dated 3 December 2004 AFMS Report 2004-5-6, Rev A dated 20 October 2004</td>
<td></td>
<td>1. FAA TCDs A10CE Plus JAA TOLs No 8 and JAA TOL No 13 apply. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.A.S.00646</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>REV. 1</td>
<td>B. 501, 551 A. 500, 550</td>
<td>Installation of single/dual Garmin GTX330D.</td>
<td>1. Installation in accordance with: ADR-2007-079-MDL-01-00 and STC MDL ADR-2007-079-MDL-01-02, or later approved Revisions. To be operated in accordance with Avionik Straubing Airplane Flight Manual Supplement Doc-No: ADR-2007-079- FMS-05.1.00 ADR-2007-079-FMS-05.2.00 or ADR-2007-079-FMS-05.3 ..</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and anything previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons ...</td>
<td>17/06/2008</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.00659</td>
<td>ZOREX S.A</td>
<td>REV. 1</td>
<td>FAIRCHILD SWERINGEN SA 226-TC</td>
<td>Installation of an L3 digital FDR F-1000 in accordance with the Engineering Order Doc.:2021094-EO-01, Issue 2 or later EASA approved Revisions. AFM Supplement Document No 2021094-AM-02 Issue 2 or later EASA approved Revisions. Maintenance Requirements and Instructions for Continuous Airworthiness:Dc. 20...</td>
<td>1. Applicable only to MF 20 SIN 397.2. This STC is approved only for the configuration as defined in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>07/12/2004</td>
<td>Active</td>
<td>10053193</td>
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<td>EASA.A.S.00685</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td>A321-231</td>
<td>VIP cabin modification in accordance with Modification Approval Sheet (MAS) ref. SOG-792 issue 1, dated March 31st, 2005</td>
<td>Airplane Flight Manual Supplement (AFM-S) ref. DTIBU/1200/39703 issue 1, dated January 18, 2005 Master Minimum Equipment List Supplement (MMEL-S) ref. &quot;01-26 P5&quot;, &quot;01-26 P6&quot;, &quot;01-33 P5&quot;, &quot;01-33 P6&quot;, &quot;01-35 P5&quot;, &quot;01-35 P6&quot;, &quot;02-52 P1&quot;, &quot;02-52 P2&quot;.</td>
<td>This STC is approved only for the product configuration as defined in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>04/06/2006</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.0069</td>
<td>SCANDINAVIAN AVIONICS DESIGN GMBH</td>
<td>REV. 1</td>
<td>HAWKER 800XP SAE 126-8008</td>
<td>Installation of an L3 digital FDR F-1000 in accordance with the Engineering Order Doc.:2021094-EO-01, Issue 2 or later EASA approved Revisions.AFM Supplement Document No 2021094-AM-02 Issue 2 or later EASA approved Revisions. Maintenance Requirements and Instructions for Continuous Airworthiness:Doc. 20...</td>
<td>This STC is approved only for the product configuration as defined in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>07/12/2004</td>
<td>Active</td>
<td>10053193</td>
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<tr>
<td>EASA.A.S.00717</td>
<td>DASSAULT FALCON SERVICE</td>
<td>REV. 1</td>
<td>MF 20G F/SN 397</td>
<td>Installation of a 2nd Air Data System</td>
<td>This STC is approved only for the configuration as defined in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>17/08/2005</td>
<td>Active</td>
<td>10043156</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.00731</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>55 / 60</td>
<td>Installation of Air Ambulance Equipment</td>
<td>AR 17/001. Component, installation, operation manual, illustrated part catalog/E Ambulance Equipment Learjet 60 “quickconversion” P/N NLJ60-25-20-20.000, CMM-L60-AAT28, CMM- L60- AAT29, 2. Airplane Flight Manual Learjet 60 Supplement, Air Ambulance Equipment P/N L60-25-20-20.000, AFM- AAT17</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked...</td>
<td>17/05/2005</td>
<td>Active</td>
<td>10042929</td>
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<tr>
<td>EASA.A.S.00734</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>550, 560</td>
<td>Installation of Air Ambulance Equipment</td>
<td>AR 13/991. Component, Installation, Operation Manual, illustrated Part Catalog/E Ambulance Equipment CITATION BRAVO “quick conversion” P/N550-25-24-19.000, CMM-CIT-AAT38 CITATION/ and ultra “quick conversion” P/N 560-25-24-16.000, CMM-CIT-4AT712, Airplane Flight Manual CITATION BRAVO Supplement Air...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” Compatibility with other aircraft/engine configuration shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked...</td>
<td>31/05/2005</td>
<td>Active</td>
<td>10424961</td>
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<td>EASA.A.S.00735</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>1900D</td>
<td>Installation of Air Ambulance Equipment</td>
<td>AR 11/991. Component, Installation, Operation Manual, illustrated Part Catalog/E Ambulance Equipment “quick conversion” P/N 10.000 Beech 1900D AirAmbulance Equipment P/N 1900D-25-20-10-000-601, CMM-BEECH-AAT432, Airplane Flight Manual Beech 1900D Supplement Air Ambulance “quickconversion” P/N 1900D...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” Compatibility with other aircraft/engine configuration shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked...</td>
<td>31/05/2005</td>
<td>Active</td>
<td>10424962</td>
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<td>EASA.A.S.00738</td>
<td>ELECTRONIQUE MAINTENANCE AVIATION</td>
<td>FAIRCCHILD SWEARINGEN SA 226-Ti</td>
<td>Installation of GPS BRNAV, EGPWS, TAS, Stormscope, Radar, MFD-Garmin CNS 530 VHF NAV GPS, Honeywell RDR 2000 Radar, BP/Goodrich WX 500 Stormscope and Avidyne EX 500 MDF installation - BF Goodrich Sky 690 TAS and Honeywell KGP 560 EGPWS provision</td>
<td>AFM Supplement “Supplement au Manuel de Vol GPS BRNAV EMA/050022 SA25L, GPS.BRNAV.EGPWS.TAS, STORMS.RDR.MFD Rev. 1Avis No. T312, F-GRN7”</td>
<td>1. This STC is limited to MSN T312 F-GRN72. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>20/02/2006</td>
<td>Active</td>
<td>10133695</td>
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<td>EASA.A.S.0074</td>
<td>SCANDINAVIAN AVIONICS DESIGN GMBH</td>
<td>HAWKER 800 XP BAE 125-800B</td>
<td>Installation of a Kannad 406 AF ELT in accordance with the Engineering Order Doc.: 2021390-EO-01, Issue 2 or later EASA approved revisionsAFM Supplement Document No. 2021390-AM-02.Issue 2 or later EASAapproved revisionsMaintenance Requirements and Instructions for Airworthiness Doc.: 2021390-CA-01</td>
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<td>Foreign/NAATC ref.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.A.S.00747</td>
<td>SCANDINAVIAN AVIONICS DESIGN GMBH</td>
<td></td>
<td></td>
<td>MU 2B-40</td>
<td>Installation of Avionics Alteration in Mitsubishi MU 2B-40</td>
<td>EASA approved Flight Manual Supplements2019747-AM-01, Issue32019747-AM-02, Issue 32019747-AM-03, Issue 32019747-AM-04, Issue 32019747-AM-05, Issue 2019747-AM-06, Issue 32019747-AM-07, Issue 2019747-AM-09, Issue 2019747-AM-10, Issue 2019747-AM-11, Issue 2 or later EASA approved issues Maintenance...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>31/05/2005</td>
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<td>EASA.A.S.00760</td>
<td>VITROCOSET S.p.A.</td>
<td></td>
<td>REV. 1</td>
<td>F27 MK 500</td>
<td>Traffic Collision Avoidance Systems (TCAS II) Installation Note: This STC supersedes EASA.A.S.00761.</td>
<td>Master Drawing List No 600-2004-000-000 Rev. IR dated 30 March 2005 1. Airplane Flight Manual Supplement No 600-2004-800-000 Rev. IR dated25 January 2005 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>1. Airplane Flight Manual Supplement No 600-2004-000-000 Rev. IR dated25 January 2005. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>07/01/2009</td>
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<td>EASA.A.S.00761</td>
<td>VITROCOSET S.p.A.</td>
<td></td>
<td>REV. 1</td>
<td>F27 MK 500</td>
<td>Traffic Collision Avoidance Systems (TCAS II) Installation Note: This STC supersedes EASA.A.S.00761.</td>
<td>Master Drawing List No 600-2004-000-000 Rev. IR dated 30 March 2005 1. Airplane Flight Manual Supplement No 600-2004-800-000 Rev. IR dated25 January 2005 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>1. Airplane Flight Manual Supplement No 600-2004-800-000 Rev. IR dated25 January 2005. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>07/01/2009</td>
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<td>EASA.A.S.00762</td>
<td>VITROCOSET S.p.A.</td>
<td></td>
<td></td>
<td>F27 MK 500</td>
<td>Radar System Replacement with Bendix King ART 2000 System. Note: This STC supersedes EASA.A.S.00762.</td>
<td>Master Drawing List No 400-0303-000-800 Rev. IR dated 29 July 2003 installer approved revisions. 1. Airplane Flight Manual Supplement No 400-0303-800-000 Rev. IR dated 4 June 2003. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>1. Airplane Flight Manual Supplement No 400-0303-800-000 Rev. IR dated 4 June 2003. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>07/01/2009</td>
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<td>EASA.A.S.00764</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td></td>
<td>CESSNA 525 &amp; CESSNA 525A</td>
<td>Installation of a Honeywell KMD 850 System and installation of a TCAS Isky.</td>
<td>- Flight Manual Supplement 2016515-AM-01, Issue 3, or laterEASA approved Revisions - Flight Manual Supplement 2016515-AM-01, Issue 3, or laterEASA approved Revisions.</td>
<td>1. This STC is approved only for the product configurations as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>13/06/2005</td>
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<td>EASA.A.S.00766</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
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<td>SAAB SF340A</td>
<td>Installation of an ADF 2 Bendix King KDF 806 System.</td>
<td>SAD &quot;Description of System and Installation&quot; Ref.: 2020789-AM-01, Issue 4 visited 22.10.2004 SAD &quot;SAAB SF340A approved Aircraft Flight Manual Supplement for ADF 2 Bendix King KDF 806&quot; Ref.: 2020789-AM-01, Issue 4 visited May 2005.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>08/06/2005</td>
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<td>EASA STC no.</td>
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<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.A.S.00767</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>A320-214</td>
<td>LIVE TV Installation</td>
<td>Modification Approval Sheet (MAS) Ref.: SOG-00-91, Issue 1 dated May 27, 2005. MPD Section 9-2 Supplement Ref.: MPD-S MOD-SOG-00-901, dated May 27, 2005</td>
<td>This STC is limited to A320-213 MSN 2165</td>
<td>04/07/2008</td>
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<td>EASA.A.S.00773</td>
<td>SCANDINAVIAN AVIONICS DESIGN GMBH</td>
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<td>BEECH 200/B200/300, B300</td>
<td>Installation of a Garmin GNS 430A/530A COMM/NAV/GPS System.</td>
<td>AFM Supplement &quot;Garmin GNS-430A/530A VHF Communications Transceiver/VOR/ILS GPS Receiver&quot; Document No. 2026288AM-02A, Issue 3 or later EASAapproved revisions. AFM Supplement &quot;Allied Signal KMR 675 Marker BeaconReceiver&quot; Doc. No. 2026288-AM-02B, Issue 1 or later EASAapproved revisions.</td>
<td>1. This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>29/06/2005</td>
<td>Active</td>
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<td>EASA.A.S.0082</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>SA227-AT, SA227-AC</td>
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<td>Installation of Artificial Horizon AIM 520-1b and EmergencyPower Supply/PS-835D.</td>
<td></td>
<td>This STC is approved only for the product configuration as defined inthe approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/06/2004</td>
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<td>EASA.A.S.0088</td>
<td>AERO ENGINEERING DESIGN LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>757-200</td>
<td>Installation of T2CAS Combined ACAS and Terrain Awareness Warning Systemin accordance with AED Design Modification Certificate 34-15-190 dated 16/07/2004 or later EASA-approved revision. Flight Manual Supplement:AED/ BT757/002</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E. InStTC and associated data shall remain valid untilsuspended, suspended or otherwise terminated. 2. If transfer of this Certificate is requested under Part 21A.116, theSTC will be reviewed. 3. This STC is approved only for the produ...</td>
<td>06/08/2004</td>
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<td>EASA.A.S.01000</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>BOEING 767-300</td>
<td>Installation of an Optical Quick Access Recorder (OQAR) according to LHT Engineering Bulletin No. 767-EB31-0213, dated 09.05.04 or later approved Revision for FOQA Purpose</td>
<td></td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other Aircraft configurations shall be determined by the installer.</td>
<td>17/08/2005</td>
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<td>EASA STC ref.</td>
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<td>LUTTHANSA TECHNIK AG</td>
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<td>BOEING 737-300</td>
<td>Installation of an Optical Quick Access Recorder (OQAR) according to LHT Engineering Bulletin No.737-EB31-0022, dated 29.09.04 or later approved Revision for FOQA Purpose.</td>
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<td>EASA.A.S.01008</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
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<td>CESSNA 525A CESSNA 525</td>
<td>Installation of Honeywell EGPWS MK V with RAAS</td>
<td>Doc. No. 12BB-MDL 12BB-EOEASA approved RUAG AFM-S Doc. No. 12BB-AFM-S. Original issue, or later approved Revision/RUAG Continued AIR procedures. Doc. No. 12BB-CAPW, Original issue, or later approved Revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other Aircraft/engine configurations shall be determined by the installer.</td>
<td>10/04/2007</td>
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<td>EASA.A.S.01014</td>
<td>BBJ DESIGN SERVICES LIMITED</td>
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<td>CAUK RESPONSIBIL ITY A320-200</td>
<td>Reconfiguration of Passenger Cabin to 180 Economy Class Passengers</td>
<td>BBJ394 Issue 1 or later approved revision</td>
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<td>17/01/2005</td>
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<td>EASA.A.S.01015</td>
<td>CESSNA DUESSELDORF CITATION CENTER GmbH</td>
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<td>CESSNA 501-551</td>
<td>Installation of a Honeywell ITA 870 and ART2000</td>
<td>Flight Manual Supplement 2903-2-8 Revision A, or later EASA approved revisions.</td>
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<td>15/06/2005</td>
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<td>EASA.A.S.01016</td>
<td>AEROTECH &amp; CONCEPT</td>
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<td>C90 A</td>
<td>Installation of Garmin GNS 430 (VHF NAV ILS GPS/IIRNAV), Allied Signal ART 2000 (weather radar), Allied Signal KMD 1250 (MFD), MD 41 (panel annunciator) and GMA 340 (markers)</td>
<td>-Dossier de modification Aerotec No 4104</td>
<td>1. Modification limited to s/n LJ 10982. EASA Supplemental Airplane Flight Manual GNS 430 must be used with this modification. 3. EASA Supplemental Airplane Flight Manual KMD 850 must be used with this modification.</td>
<td>19/01/2005</td>
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<td>EASA.A.S.01017</td>
<td>BBJ DESIGN SERVICES LIMITED</td>
<td>REV. 1</td>
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<td>CAUK RESPONSIBIL ITY 737-700, 737-800</td>
<td>Installation of 44 Passenger Seats, seats power supply and in-seat power supply system on Boeing 737-200 Reynard J2000 double seats, safety equipment, carpet and new wardrobes. Installation of General Dynamics In-seat Power System.</td>
<td>Modification Document BBJ385 Issue 4 or later approved revision</td>
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<td>24/10/2005</td>
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<td>EASA.A.S.01018</td>
<td>BBJ DESIGN SERVICES LIMITED</td>
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<td>CAUK RESPONSIBIL ITY 767-300</td>
<td>Floor Structure rework.</td>
<td>BBj Design Services Ltd Modification Summary Sheet BBJ365 dated May 18, 2005 or later EASA approved revisions.</td>
<td>1. This STC is restricted to Boeing 767-329 Registration G-VKNO. Serial Number 237652. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined...</td>
<td>08/06/2005</td>
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<td>EASA STC no.</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.01020</td>
<td>AERODATA AG</td>
<td>BEECH B200T B200 / B200T</td>
<td>Installation of TCAS 2000 with RCZ 852 Mode S Transponders</td>
<td>-AFM 076 Rev 0 or later EASA approved revision-</td>
<td>AMM-172 Rev 0</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>11/02/2005</td>
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<td>EASA.A.S.01021</td>
<td>AERODATA AG</td>
<td>B200, B300 / B200T</td>
<td>Installation of UNS 1-F Flight Management System</td>
<td>-AFM 076 Rev 1 or later EASA approved revision-</td>
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<td>11/02/2005</td>
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<td>EASA.A.S.01022</td>
<td>DASSAULT FALCON SERVICE</td>
<td>MYSTERE FALCON 20-5</td>
<td>Installation of Collins TCAS-4000 and of 2 ATC TDR 94D capable of elementary surveillance.</td>
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<td>28/04/2005</td>
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<td>EASA.A.S.01030</td>
<td>SOCIETE AIR FRANCE S.À.</td>
<td>AIRBUS A321-211 REV 1</td>
<td>Cabin reconfiguration/Change Approval Sheet ref. CAS-A320-04-015 revision 0, dated October 21, 2005</td>
<td>Change Approval Sheet ref. CAS-A320-04-015 revision 0, dated October 21, 2005</td>
<td>This STC is limited to the A321 MSN 1017</td>
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<td>08/06/2006</td>
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<td>EASA.A.S.01032</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>LEARJET 55</td>
<td>Installation of a Honeywell EGPHS mark VIII (TAWS)</td>
<td>-Engineering Order 2019360-E0-03, Rev. 2-</td>
<td>Certification Basis for this STC is CS 25 plus JAA TGL No 12 &quot;TAWS&quot;</td>
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<td>31/01/2005</td>
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<td>EASA.A.S.01035</td>
<td>E A D AEROSPACE AIRWORTHINESS</td>
<td>BEECH 1900D BEECH 1900C, AIRLINER</td>
<td>ELS/EHS mode S installation on B1900 series</td>
<td>-Top Master Drawing List 06886-JAA-MDL rev. D or later approved revision-</td>
<td>Maintenance planning document supplement DC112-0886-00A or later approved revision-</td>
<td>The Airplane Flight Manual Supplement DC107-0886-0A or later approved revisions must be used with this modification.</td>
<td>12/04/2005</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

**EASA STC No.** | **STC Holder** | **Rev.** | **Foreign/NAATC ref.** | **Model** | **Description** | **Certification basis** | **Associated Technical Documentation** | **Limitations/Conditions** | **Date of Issue** | **Status** | **System Record** |
--- | --- | --- | --- | --- | --- | --- | --- | --- | --- | --- | --- |
EASA.A.S.01037 | E A D AEROSPACE AIRWORTHINESS | BEECH 1900 |  | Installation of a Traffic Alert and Collision Avoidance System, model Honeywell CAS-67A TCASII installation on B1900 Series. | EAD Master Drawing List Document No. 0087-JAA- MDL, Revision F dated 25 May 2005, or later EASA approved revisions. Maintenance Planning Document Supplement DC112-0887-00A or later Revisions. | All wiring required for TCAS II activation must be installed. ATC Mode S Transponders must be installed. | 18/07/2005 | Active | 10176027 |

EASA.A.S.0104 | ATS INTERNATIONAL | REV. 1 | B737-700/-800/-900 B737-300/-400/-500/-600 | Installation of the enhanced Mode S surveillance in accordance with Master Data List reference 0159-01-A-2103-R03 or later EASA-approved revision. | 1. The AFM Supplement reference No 0159-01-A-2301- R01 is a required part of this STC. 2. Installation of the enhanced Mode S surveillance in accordance with Master Data List reference 0159-01-A-2103-R02 or later EASA approved revision.3. This STC is approved only for the product configuration as defined... | 16/12/2004 | Active | 10043086 |

EASA.A.S.01041 | UNI AR ENTREPRISE SABUNI AIR ENTREPRISE | BEECH C90A | Sealed Lead Acid Battery. | - FAA STC no. SA00257AT - United Beechcraft Master Drawing List MDS LF/32/101 (15/06/1993). FAA Form 337 No MLR600L. | 1. Supplemental Airplane Flight Manual No UA502 (French language) must be used with this modification. 2. Modification limited to Beechcraft C90A sn LJ-1361. 3. The STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". C... | 11/04/2005 | Active | 10014004 |

EASA.A.S.01042 | EADS SOGERMA | AIRBUS A340-300 | Installation of an upper class with side facing Britax Malcom seats with inflatable restraints, monument, bar and chandelier, and modification of the premium and economy class by replacing Note: This revision reflects the transfer of this STC from EADS Sogema Services to Sogema Services. | MAS BDD-00-1000 issue dated 1/08/2005 Master Data List Drawing No. MDS-00-1000 issue No. A or later EASA approved revision. Flight Manual Supplement page 1.02.00 dated 01/07/2005 Technical repercupaction sheet Ref. 05-1006A. | 1. This STC is limited to Airbus A340-300 MSN 214, 2002, 003, 164 and 225. A copy of this STC and of the MAS/MDL must be enclosed in the files of each modified aircraft. | 07/01/2008 | Active | 10014005 |


EASA.A.S.01049 | FR AVIATION LTD/BA COBHAM AVIATION SERVICES UK | CAA UK RESPONSIBILITY | 228-202K | Introduction of ACAS and Mode S | - Design Change Definition Document DC-0000 issue B or later approved revision. | This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph "Description". Compatibility with other aircraft/engine configurations shall be determined by the installer. | 27/06/2006 | Active | 10043010 |

EASA.A.S.01051 | HAWKER BEECHCRAFT LTD | HAWKER HS 125-700B | Increase Maximum Zero Fuel Weight | Instructions for Continued Airmenfitness ACS - R-007-711422-Weight Statement CL/WEI /800. | | | 22/02/2005 | Active | 10014008 |
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<th>Foreign/NAA STC ref.</th>
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<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td>10014009</td>
<td>SCANDINAVIAN</td>
<td>CESSNA 525-525A</td>
<td>Installation of a Dual MST 67A Mode S Transponder System.</td>
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<td>13/06/2005</td>
<td>Active</td>
<td>1014162</td>
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<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td>10014014</td>
<td>SCANDINAVIAN</td>
<td>CESSNA 525</td>
<td>Installation of Dual Garmin GTX 330D Mode SAFM Supplement 2004-16-8 Revision A or later EASA approved Revisions.</td>
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<td>17/11/2005</td>
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<td>PA.28RT-201T (TURBO ARROW IV)</td>
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<td>Installation of GARMIN GNS 430 (VHF NAV/ILS/GPS) and MD 41 (panel annunciator) Removal of VHF K Y 97A, GPS 150XL, NAV/ILS 351 and Glide GLS 350. Installation of GNS 430 and MD 41 panel annunciator, auto-pilot coupled.</td>
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<td>AMD/BA FALCON 10</td>
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<td>Installation of Rockwell Collins TCAS-4000 incl. Dual Mode S Transponder TDR-94D in accordance with the Engineering Order Doc 2022256-EO-01 Issue 2 or later EASA approved revisions</td>
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<td>Installation of a Kannad 406 ELT in accordance with Engineering Order Doc 2022256-EO-03, Issue 1 or later EASA approved revisions</td>
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<td>Installation of a Honeywell Mark V EGPWS soft 217-217 windshield display in the Collins EFIS 86-C14</td>
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<td>Installation of TCAS II system in accordance with MDL BCA 70-00 rev. 3 dated 18/31935</td>
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<td>-AFM Supplement ref.: AFM-25-886 Iss. A</td>
<td>-AFM Supplement Doc.-No.: OMB-AFMS, Rev. N/Cor later EASA approved revision</td>
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<td>EASA.A.S.01077</td>
<td>EADS SOGERMA</td>
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<td>AIRBUS A330-203</td>
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<td>Qatar Airways - Cabin Modification. Modification number: SOG-25-886 issue 1. Note: This revision reflects the transfer of this STC from EADS Sogema Services to Sogerma Services.Model: Airbus A330-203 MSN 463 and 564</td>
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<td>BEECH 1900D</td>
<td>Emergency Transmitter Kannad 406 MHz installation on Beechcraft 1900 D.</td>
<td>- BCA No. 81 Modification File - BCA/RDO/1404/04 Drawing (Merely ELT installation) - BCA/RDO/030904 Drawing (ELT and Programming Unit)</td>
<td>1. This STC is limited to Beechcraft 1900 am U.E.-1 and up.2. Required Pre-Maintenance and Continued airworthiness...</td>
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<td>BEECH B200C</td>
<td>Installation of a Universal TAWS system and related components.</td>
<td>- EASA approved AFM Supplement Doc. 2021128-AM-01, issue 5 or later EASA approved issue - Description of Change 2021128-DE, Issue 3</td>
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<td>BEECH B200</td>
<td>Installation of GPS TNL-2101 I/O APPR+ Systems.</td>
<td>-AFM Supplement 2021128-AM-02 Issue 1 or later EASA approved revision - Continued Airworthiness Instructions 2021128-CA-02</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>BEECH B300C</td>
<td>1. Installation of Dual Mode S Transponder MST-67A System with ELS capabilities (installation prior to 31-Jan-2007). Installation of Dual Mode S Transponder MST-67A System with ELS and EHS capabilities (installation on or after 31-Jan-2007)</td>
<td>1. For installations prior to 31-Jan-2007:- AFM Supplement 2021128-AM-03, Issue 1 or later EASA approved revision - Continued Airworthiness Instructions 2021128-CA-03.2. For installations on or after 31-Jan-2007:- Engineering Order, Doc. No. 2021128-EO-03, Issue 5, dated 31-Jan-2007.</td>
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<td>AERO-DENST GMBH &amp; CO. KG</td>
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<td>Installation of a Cabin Entertainment System with Airshow 4X0.</td>
<td>Aerodienst engineering Order N° 203-LR, Rev. IR, or later EASA-approved revisions.</td>
<td>For certification basis refer to FAA TCDS A10CE plus FAR 25.1309 at Amdt. 25-41 and FAR 25.1309 at Amdt. 25-54 and plus/or JAA TGL No.17, IFE Systems</td>
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<td>MYSTERE FALCON 50</td>
<td>Installation of a Honeywell Mark VII EGPWS soft 217-217 with terrain display in the MFD</td>
<td>-Change Approval Sheet DFS 5303 issue original dated February 8th 2005 -AFM Supplement DFS No. 56 issue original dated February 8th 2005 if the aircraft has steep approach capability AFM Supplement DFS No. 51 issue original dated February 8th 2005</td>
<td>This STC is only valid for Falcon 50 equipped with a Universal FMS 512B with GPS-1500, or UNS-1C/1C+/1D/1E/1F/1K+/1L and with Collins EFIS-86C(x)/weather radar WMR-350</td>
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<td>PA-31</td>
<td>Installation of Avidyne FM850, 2x Garmin GTX330D Mode S Transponder and Garmin GMA340 Marker/Audio</td>
<td>EASA approved AFM Supplement Avidyne FM850 Flight Display LBA2004-024-01 Rev A Garmin GTX330D Mode S Transponder LBA2004-024-00 Rev A Garmin GMA340 Marker Audio LBA2004-024-02 Rev A or later EASA approved issues</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
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<td>Installation of Mode S Transponder TDR94D with new Top Antenna.</td>
<td>AFM Supplement Mode S Transponder Collins TDR 94 D: Doc. No.: ACG 2004-014-00 Rev A, Engineering Order ACG2004-014-00</td>
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<td>737-300, 737-400, 737-500</td>
<td>Installation of an Artex Emergency Locator Transmitter</td>
<td>- ATS International Master Data List No 0233-01-A-2103 revision 2 - Airworthiness Limitations as specified in ATS - Airworthiness Data Sheet Supplement No 0233-01-A-2203 revision 52.6.5</td>
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<td>CESSNA 550 BRAVO</td>
<td>Upgrade of GPWS to EGPWS</td>
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<td>Installation of WX-500 Stormscope System</td>
<td>EASA approved Flight Manual Supplement 2019974-AM-02, issue 2, or later EASA approved revision.</td>
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<td>Installation of BendixKing KLN 94 GPS System.</td>
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<td>GULFSTREAM AEROSPACE Ltd. CAA UK RESPONSIBILITY</td>
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<td>GULFSTREAM IV</td>
<td>Installation of Artex C406N Emergency Locator Transmitter</td>
<td>Definitions Document C406/0138/001 Issue 3 dated 25 January 2015 or later approved revision.</td>
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<td>A330-200 Air Algerie Quick Change VIP Kit OnePlus.</td>
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<td>Introduction of Honeywell Mk VII EGPIWS</td>
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<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>STC TWENTY ONE LIMITED</td>
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<td>CAP 232</td>
<td>Introduction of Rudder, Throttle and Brake Controls for Hand Operation by Paraplegic Pilot with upper Body Strength only</td>
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<td>1. Prior to installation of this modification the installer must determine that the interaction between this modification and any other previously installed modification will introduce no adverse effect on the airworthiness of the product. 2. Serial Number 11 only.</td>
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<td>B737-700 &amp; B737-800</td>
<td>Introduction of SSR Mode S Enhanced Surveillance</td>
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<td>Certification Basis: applicable requirements of CS25 plus JAA TGLs No 8 and 13.1</td>
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<td>FALCON 10 DFS 1261 dtd 13.Oct 2005: Installation of a TAWS Honeywell Mark VIII EGPRS soft 011 see Change approval sheet DFS 1261-APM Supp F10 DFS 29 toDTM 721(A) with DFS 1261 incorp &quot;TAWS Honeywell Mark VIII Soft 011&quot;-APM Supp F10 DFS 31 to DTM 721 (A) with DFS 1261 incorp &quot;EGPRS Steep Approach Function&quot;-Wiring Diagram DFS 10 860 31-56-01 XAnd associated list of affected technical docu...</td>
<td>1. This STC is approved only for the configuration as defined in Mod Sheet DFS 1261.2. Compatibility with other aircraft/engine configurations shall be determined by the installer</td>
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<td>ATS INTERNATIONAL</td>
<td>767-200,-300 MK V EGPRS system installation.</td>
<td>Airplane Flight Manual Supplement (AFMS) ref. 0203-01-A-2301 revision 0 dated June 13, 2005 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determ..., the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>BOEING 767-200-300 Installation of Elementary Surveillance Dual ATC Mode S on B767-300The ATC Mode S computers TPR 720 are replaced by a TPR D01 which is compliant with ATC Mode S Elementary surveillance requirements.ATC Mode S Elementary surveillance implementation is described in Air France Service Bulletin No. AF S...</td>
<td>Air France Change Approval Sheet No. CAS-767-04-001 dated December 1, 2004 Air France Service Bulletin No. AF SB 8767-04-01134-001 &quot;EATC Installation on 8767-300&quot;</td>
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<td>BOEING 767-200-300</td>
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<td>Air France Change Approval Sheet No. CAS-767-04-001 dated December 1, 2004. Air France Service Bulletin No. AF 88 B767-04-001-34-001 “EATC Installation on B767-300”</td>
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<td>Installation of EGPWS. This STC supersedes and cancels DGAC STC IM29-SF-0282.</td>
<td>BOEING 747-200-300</td>
<td>Installation of EGPWS. This STC supersedes and cancels DGAC STC IM29-SF-0282.</td>
<td>Group 1 (Air France Fleet): AEO 47-34-0715. Group 2 (Customer Fleet): CAS B747-04-010. AFM Supplements D6-33747-228, D6-33747-2B3 and D6-33747-3B3 are applicable to Air France B747-200/300 aircraft. AFM Supplement No. MA.PM0006/05 is applicable to Pullmantur Air aircraft.</td>
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<td>BOEING 747-300</td>
<td>Elementary surveillance dual ATC Mode S installation.</td>
<td>BOEING 747-300</td>
<td>Elementary surveillance dual ATC Mode S installation.</td>
<td>Change Approval Sheet ref. CAS B747-04-009 Revision 0, dated February 10, 2005 or later approved revision.</td>
<td>This STC is limited to the B747MSN 24106.</td>
<td>15/03/2007</td>
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<td>EASA.A.S.01144</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>CESSNA 560 (CITATION V)</td>
<td>Installation of ACAS II (ACSS TCAS 2000, CH7) with Dual Mode S Transponders RCZ-833J ELS.</td>
<td>CESSNA 560</td>
<td>Installation of ACAS II (ACSS TCAS 2000, CH7) with Dual Mode S Transponders RCZ-833J ELS.</td>
<td>- Master Document List, Dok Nr. 093C-MDL, Rev N/C dated 03.03.04- AFM Supplement, Dok Nr. 093C-AFM, Rev N/C dated 03.03.04-Continued Airworthiness Plan (Maintenance Manual Supplement), Dok Nr. 093C-CAWP, Rev N/C dated 03.03.04</td>
<td>10/03/2005</td>
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<td>EASA.A.S.01147</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>PA-31T</td>
<td>Installation of radio navigation and communication systems in accordance with Document Reference 5304 - Revision 2 dated April 15, 2005.</td>
<td>PA-31T</td>
<td>Installation of radio navigation and communication systems in accordance with Document Reference 5304 - Revision 2 dated April 15, 2005.</td>
<td>AFMS referenced 5304. STC Document Reference 5304 Revision 2 dated April 15, 2005</td>
<td>This STC is limited to Piper PA-31T SN 800037.</td>
<td>11/10/2005</td>
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<td>EASA.A.S.01148</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>650</td>
<td>Installation of Honeywell CAS-67 ACAS II and Dual MTI 87A Mode-S Transponder.</td>
<td>650</td>
<td>Installation of Honeywell CAS-67 ACAS II and Dual MTI 87A Mode-S Transponder.</td>
<td>- Airplane Flight Manual Supplement for Cessna 650 Citation III, Doc. No C650-01-0601 Revision 0 dated 17.03.2005 or later approved revision.- Instructions for Continuous Airworthiness, Doc. No C650-01-0701, Revision 0 dated 17.03.2005 or later approved revision.</td>
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<td>EASA.A.S.01149</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>650</td>
<td>Installation of Allied Signal Solid State Flight Data Recorder.</td>
<td>650</td>
<td>Installation of Allied Signal Solid State Flight Data Recorder.</td>
<td>- Airplane Flight Manual Supplement for Cessna 650 Citation III, Doc. No C650-02-0601 Revision 0 dated 03.03.2005 or later approved revision.- Instructions for Continuous Airworthiness, Doc. No C650-02-0701, Revision 0 dated 03.03.2005 or later approved revision.</td>
<td>11/04/2005</td>
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## List of EASA Supplemental Type Certificates

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<td>Installation of Allied Signal AR120 Solid State Cockpit Voice Recorder.</td>
<td>-Airline Flight Manual Supplement for Cessna 650 Citation III, Doc. No. C650-03-0880 Revision 0, dated 12.03.2005 or later approved revision. -Instructions for Continuous Airworthiness, Doc. No. C650-03-0701, Revision 0, dated 12.03.2005 or later approved revision.</td>
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<td>EASA.A.S.01151</td>
<td>SOCIETE AIR FRANCE S.A.</td>
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<td>AIRBUS A320</td>
<td>Cabin passenger seats relocation and floor proximity emergency escape path marking change</td>
<td>Change approval sheet No CAS-A320-05-021 revision 1, dated March 22, 2005</td>
<td>-Airline Flight Manual Supplement for Cessna 650 Citation III, Doc. No. C650-03-0880 Revision 0, dated 12.03.2005 or later approved revision. -Instructions for Continuous Airworthiness, Doc. No. C650-03-0701, Revision 0, dated 12.03.2005 or later approved revision.</td>
<td>07/04/2005</td>
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<td>EASA.A.S.01154</td>
<td>AVIATION DEFENSE SERVICE</td>
<td>REV. 2</td>
<td>B200 / 200</td>
<td>B200 / 200</td>
<td>Installation of an Emergency Locator ELT 96-406 MHz.</td>
<td>Modification AIV-MB200-23-046A, Supplemental AFM No. 1, 3, &amp; 6.</td>
<td>1. This STC requires a pre-modification: 2 bottom transponder antennas. 2. This STC is limited to Beechcraft 900 and Beechcraft 200 s/n L808.</td>
<td>15/03/2005</td>
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<td>EASA.A.S.01156</td>
<td>AVIATION DEFENSE SERVICE</td>
<td>C90, E90</td>
<td>C90, E90</td>
<td>C90</td>
<td>Installation of an Emergency Locator ELT 96-406 MHz.</td>
<td>Modification AIV-MB200-34-074A, Supplemental AFM No 2 &amp; 3.</td>
<td>1. This STC requires a pre-modification: 2 bottom transponder antennas. 2. This STC is limited to Beech s/n BB199 and Beech 200 s/n BB439.</td>
<td>28/07/2005</td>
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<td>EASA.A.S.01157</td>
<td>AVIATION DEFENSE SERVICE</td>
<td>C90</td>
<td>C90</td>
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<td>Installation of Mode S Transponder MST 67A.</td>
<td>Modification AIV-MB200-34-074A, revision a dated 20 June 2005 or later EASA approved revision.</td>
<td>1. This STC requires a pre-modification: 2 bottom transponder antennas. 2. This Modification is limited to Beechcraft C90 s/n LJ860.</td>
<td>22/07/2005</td>
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<td>EASA.A.S.01161</td>
<td>DASSAULT FALCON SERVICE</td>
<td>REV. 1</td>
<td>DFS 10</td>
<td>DFS 10</td>
<td>Installation of two FMS UNS-1L Full Motion Simulators B737 &amp; P-8760</td>
<td>see Change approval sheet DFS 1255-APM Supp F10 DFS 32 to T72(A) &quot;Two FMS UNS-1L Operation / DFS 1255 incorporated&quot;-Wiring Diagram DFS 10 600 34-56-01 &amp; DFS 10 600 34-56-11and associated list of affected technical documents</td>
<td>1. This STC is approved only for the configuration as defined in the approved design data referred to in the paragraph “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>02/06/2006</td>
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<td>EASA.A.S.01163</td>
<td>DASSAULT FALCON SERVICE</td>
<td>REV. 1</td>
<td>DFS 10</td>
<td>DFS 10</td>
<td>Installation of a Honeywell MK-VI GPWS &amp; Honeywell RT 300 Radio Attenuator</td>
<td>see Change approval sheet DFS 1255-APM Supp F10 DFS 30 to T72(A) with DFS 1252(concernant utilisation de l’avion equipe d’EVI UNIVERSAL 550/560&quot;-Wiring Diagram DFS 10 600 24-41-01, 34-21-10, 34-21-20 and associated list of affected technical documents</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” and “Associated Technical Documentation”. Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>11/01/2007</td>
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<td>EASA.A.S.01165</td>
<td>ATLANTIC AEROSPACE &amp; ENGINEERING LIMITED</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>JETSTREAM 3100 SERIES</td>
<td>Installation of a Honeywell MK-VI GPWS &amp; Honeywell RT 300 Radio Attenuator</td>
<td>-Atlantic Aerospace &amp; Engineering Definition Document DDI AAE/4/1 Issue 2 dated March 10 2005 or later approved revision-AAE Flight manual Supplement AAE II Issue 1 or later approved revision</td>
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<td>06/04/2005</td>
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<td>EASA A.S.01167</td>
<td>E.I.S.A.</td>
<td>REV 1</td>
<td></td>
<td>FALCON 10</td>
<td>Installation of a TCAS 4000 unit from Rockwell/Colins/Installation of a TCAS 4000 unit from Rockwell/Colins</td>
<td>- Data List EISA 08 SDL 0585 Rev. 7 - AFM supplement EISA 06AFM 0585 Rev. 4 - Antenna Installation according to the EISA/STC no. EASA A.S. 0070 - Data List EISA 08 SDL 0585 Rev. 7 - AFM supplement EISA 06AFM 0585 Rev. 4 - Antenna Installation according to the EISA/STC no. EASA A.S. 0070</td>
<td>30/06/2006 Orphaned</td>
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<td>EASA A.S.01168</td>
<td>EADS Sogerma</td>
<td>REV 1</td>
<td></td>
<td>AIRBUS A330-243</td>
<td>Cabin configuration for Gulf Air. All documents related to this STC are listed in the Modification SOG-25-1075 iss. 01 (or later EASA-approved revision). Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services.Model: Airbus A330-243 MSN 276, 281, 286, 287, 3...</td>
<td>Modification SOG-25-1075 iss. 01 (or later EASA-approved revision)</td>
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<td>07/01/2008 Active</td>
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<td>EASA A.S.01169</td>
<td>EADS Sogerma</td>
<td>REV 1</td>
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<td>AIRBUS A310</td>
<td>Installation of an ATC Mode S EHS, ATC mode S modification for compliance with Enhanced Surveillance. Replacement of the Rockwell/Colins TRP-720 ATC by TRP-901 ATC. Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services.</td>
<td>Modification Approval Sheet SOG-34-866 Issue 02: Aircraft Flight Manual Supplement: AFM-34-52-00 Issue A dated 15/07/2004. Compliance CS-34-866-01 Issue A.</td>
<td>Supplemental Type Definition Document No STG25-0300M, Issue 01/Rev. 1 or later approved Revisions.</td>
<td>07/01/2008 Active</td>
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<td>EASA A.S.01171</td>
<td>Lufthansa Technik AG</td>
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<td>BOEING 777-300</td>
<td>Installation of Power Control Systems and Data Outlets on the existing B/C Seats and installation of new ECO Class Seats with new IFF and Power Control Systems/Data Outlets on Boeing 777-300 A/C of Cathay Pacific Airways.</td>
<td>- Supply Engineering Design Flight Manual Supplement EAD/34-52-00 Issue A or later approved revision</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>01/03/2005 Active</td>
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<td>EASA A.S.01173</td>
<td>Aero Engineering Design Ltd.</td>
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<td>BAE HS 748</td>
<td>Installation of a Honeywell MK VI EGPWS</td>
<td>- JED Design Modification Certificate 34-23-223 dated February 10, 2005 or later approved revision-Aero Engineering Design Flight Manual Supplement AED/34-52-00 Issue 1 or later approved revision</td>
<td>02/03/2005 Active</td>
<td>1014081</td>
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<td>EASA A.S.01174</td>
<td>Bournemouth Aviation</td>
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<td>BOEING 737-200</td>
<td>Revised Seating Layout of 124 Seats and 130 Seats using BAC1288 Series Seats</td>
<td>- Bournemouth Aviation Consultants Compliance Document No BAC/737/005 dated March 3 2005 or later approved revision</td>
<td>17/03/2005 Active</td>
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<td>Bournemouth Aviation</td>
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<td>BOEING 737-200</td>
<td>Deactivation of Type III Exits &amp; Installation of a 60 Seat VIP Passenger configuration.</td>
<td>- Bournemouth Aviation Consultants Compliance Document No BAC/737/005 Top Drawer BAC 1289-01</td>
<td>03/03/2005 Active</td>
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<td>JET AVIATION VIENNA GMBH</td>
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<td>550</td>
<td>Integration of a TCAS and Upgrade of Mode S Transponders to Elementary Surveillance</td>
<td>- 5970/AT-112 Modifications for Continued Airworthiness - 5970/AT-1-06 Airplane Flight Manual Supplement</td>
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<td>EADS SOGERMA REV 1</td>
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<td>DC-9-21</td>
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<td>ELT Kannad 406 AF modification (SOQ-25-110) (Note: This revision reflects the transfer of this STC from EADS Sogarma Services to Sogetia Services.)</td>
<td>Airplane Flight Manual Supplement ref FMS 25-110; issue D, dated November 11, 2005</td>
<td>1. This STC is limited to the D-9 MSN 4708</td>
<td>04/01/2008</td>
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<td>BEECH C90 A</td>
<td>BEECH C90</td>
<td>Installation of Weather Radar RDR-2000</td>
<td>- Modification File No 50/BCA - rev 2 or later approved revision</td>
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<td>Modification limited to Beech C90 S/N LJ-684 and Beech C90/SN LJ-1526</td>
<td>17/03/2005</td>
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<td>EASA.A.S.01192</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
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<td>A300F4-608ST</td>
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<td>STC ST46MOD001 Issue 01: A300-600ST Data Link System. This controller data link communication based on VDL mode 2 project is led by Eurocontrol. The crew will be able to exchange data and messages with a ground-based station airline operation centre and air traffic services.</td>
<td>Change Approval Sheet No ST46MOD001 Issue 01: Supplement Aircraft Flight Manual A300-600ST, dated 24/03/2005</td>
<td>This STC is approved only for A300F4-608ST.</td>
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<td>12/04/2005</td>
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<td>Installation of Universal Terrain Awareness Warning System (TAWS)</td>
<td>- Engineering Order Doc. 2024352-EO-01 Issue 2 or later EASA approved revisions-AM Supplement Document No 2024352-AM-01 Issue 2 or later EASA approved revisions</td>
<td>- Continued Airworthiness Document No 2024352-CA-01 Issue 1</td>
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<td>EASA.A.S.01194</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>BAE ATP JETSTREAM SERIES 6100</td>
<td>Installation of Universal UNS-1L Flight Management System</td>
<td>-Engineering Order Doc. 2024346-EO-02 Issue 2 or later EASA approved revision-</td>
<td>AFM Supplement Document No 2024346-AM02 Issue 2 or later EASA approved revision-Continued Airworthiness Document No 2024345-CA-02</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.A.S.01199</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
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<td>550, 560</td>
<td>Installation of Allied Signal SSU/FDR</td>
<td>Master Document List C550-122401-0107 Rev. 0 or later approved RevisionFlight Manual Supplement C550-122401-0001 Rev. 0 or later approved Revision</td>
<td>Certification Basis of the Change: FAR 25 as defined in FAA305 1020E Rev. 44</td>
<td>19/04/2006</td>
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<td>FALCON 10</td>
<td>Installation of Collins ATC with flight ID and Gables Control Panel</td>
<td>-Change Approved Sheet DFS 1260 issue original dated 31 March 2005.</td>
<td>STC applicable only to Falcon 10 aircraft equipped with - 2Collins ATC TDR-94D p/n 622-9210-001/002/003 or 004 - and ATC control box Gables G7 130-XX</td>
<td>11/04/2005</td>
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<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
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<td>SOCATA TB 21</td>
<td>Installation of MTV-9-B/198-52 hydraulically controlled 3-blade constant speed propeller</td>
<td>-Instructions for Continued Airworthiness and Installation Instructions No E-1189-Airplane Flight Manual Supplement No E-1157-German “Verhang zum Flughandbuch” No E-1158</td>
<td>01/06/2005</td>
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<td>AERO-DIENST GMBH &amp; CO. KG</td>
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<td>60</td>
<td>Installation of an Iridium Satellite ICS-200 System.</td>
<td>Aerodienst Engineering Order No 200-ER-1, Rev. IR, or later EASA-approved revisions.</td>
<td>For certification basis refer to FAA TCDS A110E plus FAR 25.1399 at Amdt. 25-41 and FAR 25.1529 at Amdt. 25-54.</td>
<td>28/04/2005</td>
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<td>AERO-DIENST GMBH &amp; CO. KG</td>
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<td>LEARJET 60</td>
<td>Installation of a SSFDR F1000 System.</td>
<td>Aerodienst Engineering Order No 197-ER-1, Rev. IR, or later EASA-approved revisions.</td>
<td>1. The certification basis refers to FAA TCDS A10CE plus FAR 25.1399 at Amdt. 25-41 and FAR 25.1529 at Amdt. 25-54. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine config...</td>
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<td>EASA.A.S.01207</td>
<td>HAWKER BEECHCRAFT Ltd.</td>
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<td>BAE 125 SERIES 800B</td>
<td>Raytheon Aircraft Services Modification SC0159R. Runway Awareness &amp; Advisory System (RAAS) to the MK VII ESPHS Computer</td>
<td>RAS Certification Plan ASC-C-008-JAV1499 Issue 1 dated February 2005 or later approved revision-RAS AFM Supplement 16 issue 1 to AFM ref HS 1.10</td>
<td>21/06/2005</td>
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<td>EASA.A.S.01208</td>
<td>HAWKER BEECHCRAFT Ltd.</td>
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<td>HAWKER HS 125-700B</td>
<td>Installation of a Collins AHRS-3000A Attitude Heading Reference System.</td>
<td>Certification Plan ASC- C-007-AV1518 Issue 1, dated February 22, 2005 or later EASA approved revisions-Change SC0194U dated May 20, 2005 or later EASA approved Revisions</td>
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<td>HAWKER HS 125-700B</td>
<td>Installation of Traffic Alert &amp; Collision Avoidance System (TCAS-4000), Mode B Elementary Surveillance (Flight ID) and Emergency Locator Transmitter (Artex C406-2)</td>
<td>Certification Plan ASC-C-007-AV1518 Issue 1 Dated February 18 2005 or later approved revision</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>BEECH 95-555</td>
<td>Radio and navigation change with BRNAV capability and STEC 55X automatic pilot steered</td>
<td>Modification File No. AEROTEC STC0555</td>
<td>Airplane Flight Manual Supplement No Modification 055, edition 1, revision 0, dated 31st March 2005 or later approved revision</td>
<td>Modification limited to s/n TC1143AFMS No modification 055 must be on board</td>
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<td>BEECH E90</td>
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<td>EADS SOGERMA REV 1</td>
<td>ATR 42-320 SERIES</td>
<td>Pseudo-Loess configuration, RESSAC configuration and RAJ configuration Note: This revision reflects the transfer of this STC from EADS Sogema Services to Sogema Services Model/Aerospaliate ATR 42-320</td>
<td>AFM Supplement for Pseudo-Loess configuration referred 238045/04/D0DEV Issue 7, AFM Supplement for RESSAC configuration referred 238045/04/D0DEV Issue 2, AFM Supplement for RAJ configuration referred 240045/04/D0DEV Issue 2</td>
<td>Weight &amp; Balance Supplement for Pseudo-Loess, RESSAC and RAJ configurations refer...</td>
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<td>ELT Kannad 406 AF Modification Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services. Model: Aeronautica ATR 42-320 MSN 078</td>
<td>Technical Repercussion Sheet referred SGG-25-1104/Master Drawing List MD1-1104, Issue No. A dated February 2, 2005</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>Falcon 10</td>
<td>DFS 1251 std. 2.6.05: Installation of a Multi functional display MFD 640</td>
<td>see Change approval sheet DFS 1251-Wiring Diagram DFS 10 603 34-121 and associated list of affected technical documents</td>
<td>1. This STC is approved only for the configuration as defined in Mod Sheet DFS 1251. Compatibility with other aircraft/ engine configurations shall be determined by the installer</td>
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<td>EASA.A.S.01226</td>
<td>Dassault Falcon Service</td>
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<td>Mystere Falcon 900</td>
<td>Incorporation of RAAS function on EGPSWS soft 222-222</td>
<td>Change Approval Sheet DFS 9470 CAS issue Original dated April 14, 2006Airplane Flight Manual DTM 20103 DFS Supplement n°22 dated June 14, 2006</td>
<td>Applicable to MF900 equipped with EGPSWS Mark V soft 222-222 Installed with DFS 9472 STC</td>
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<td>Beech C90 A</td>
<td>Skywatch SKY1497 TCAS I installation on Beech C90A</td>
<td>Modification file No. 68/BCA rev 0 of 15th February 2005</td>
<td>1. AFMS No. 68/BCA rev 0 of April 2005 must be used when modification is installed. Modification is limited to Beech C90A S/N LJ-1526</td>
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<td>BOEING 747-400</td>
<td>Flight Management System (FMS).</td>
<td>Aeronautical Design Modification Certificate</td>
<td>DFS 1250-Wiring Diagram</td>
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<td>BOEING 727-200</td>
<td>Installation of Honeywell Mk V3 EGPWS.</td>
<td>Design Modification Certificate 34-12-237 Issue 3</td>
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<td>HS 125 SERIES 700B</td>
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<td>MYSTERE FALCON 20-5</td>
<td>Installation of one FMS Universal UNS-1L SOFT 802.8 X in accordance with the</td>
<td>Aircraft Flight Manual Supplement AEGST7571904</td>
<td>Aircraft Flight Manual Supplement AEGST7571904</td>
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**List of EASA Supplemental Type Certificates**

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<td>LEARJET 55</td>
<td>Installation of Secondary UNS 1F</td>
<td>Engineering Order 2005-3, Rev. A, or later EASA approved revisions; AFMS 2005-3, Rev. A, or later EASA approved revisions</td>
<td>Refer to AFM Supplement Ref. DC113-0846 Rev B and to Maintenance Planning Document Supplement DC114-0846-00 rev A.</td>
<td>This STC is limited to Falcon 2000 S/N 0123. This STC is approved only for the configuration as defined inmod DSS0566.</td>
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<td>Installation of ELT 406</td>
<td>Engineering Order 2005-2, Rev. A, or later EASA approved revisions; AFMS 2005-2, Rev. A, or later EASA approved revisions</td>
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<td>SOGERMA SERVICES</td>
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<td>AIRBUS A320-211</td>
<td>Cabin layout change from 144 to 155 pax configuration in accordance with the Master Drawing List ref. MDL-25-851, Issue B, dated August 22, 2004 or later EASA approved revision. Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services.</td>
<td>Top Master Drawing List 06846-JAA-MDL Rev. D. AFM Supplement DC113-0846 Rev. B. Maintenance Planning Document Supplement DC114-0846-00 rev A.</td>
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<td>EAD AEROSPACE</td>
<td>A300/30</td>
<td>A300/310</td>
<td>ESPWS Installation on Airbus A300A310</td>
<td>Engineering Order 2005-3, Revision A, dated 22 March 2005 or later EASA approved revisions.</td>
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<td>AFM Supplement DRS066-DFS AFM Supplement F20003 to STM 53 and associated list of affected technical documents</td>
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<td>BEECH 8300</td>
<td>BEECH 8300/8200</td>
<td>Installation of Dual GTX 330D Mode S Transponder System.</td>
<td>AFM Supplement Dual GTX 330D Mode S Transponder Doc. No. 26262867-AM-01 Issue 3 or later EASA approved revisions.</td>
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<td>DFS06608 Rev. 1, October 05: Airview Communication and Security System</td>
<td>see Change approval sheet DFS200608DFS AFM Supplement 20003 to STM 53 and associated list of affected technical documents</td>
<td>Engineering Order 2005-4, Rev A dated 22 March 2005 or later EASA approved revisions.</td>
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<td>BOEING 767-300</td>
<td>BOEING 767-300</td>
<td>Monarch Aircraft Engineering Modification MON/767/295/2965 Installation of a 38 Premium Class Cabin</td>
<td>Monarch Aircraft Engineering Modification MON/767/295/2965</td>
<td>Subject AFM supplement is associated to Dassault Aviation Mod M1832 &quot;Installation d'un hublot inférieur avec volet de protection&quot; applicable to F20 DF SN 397 only</td>
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<td>MYSTERE FALCON 200</td>
<td>MYSTERE FALCON 200</td>
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<td>HAWKER BEECHCRAFT LTD.</td>
<td>BAE 125 SERIES 800B</td>
<td>Special Interior Refurbishment</td>
<td>Certification Point ASC-C-008-AW1476 Issue 1 Dated February 2005 or later approved revision - Instructions for Continued Airworthiness ASC-C-008-AW1484 Issue 1 dated April 18, 2005 or later approved revision - Change number. SC0246A.</td>
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<td>BOEING 767-300</td>
<td>B-767- Cabin Layout Change - Air Europe</td>
<td>Supplemental Type Definition Document Doc. No. STOD 00-05/03M Issue 1. Instructions for Continued Airworthiness Doc. No. ICA 25-1009-00 dated May 30, 2005 or later approved issue</td>
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<td>FR AVIATION Ltd./BAE COBHAM AVIATION SERVICES UK</td>
<td>FALCON 20</td>
<td>FR Aviation Modification Reference DC3002 Introduction of Flight Manual Supplement FRAFM/106</td>
<td>Approved Flight Manual Reference DTM/188 FR Aviation Flight Manual Supplement FRANE FNE155 Revision 0 dated June 2005 or later approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/cabine configurations shall be determined by the installer.</td>
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<td>AIM AVIATION (HENSHELLS) LTD</td>
<td>AIRBUS A300-600</td>
<td>Installation of a Galley bulkhead.</td>
<td>AIM Aviation (Henshalls) Definition Document DD 1011 Issue New, dated 12 April 2005 or later EASA approved revisions.</td>
<td>1. Viewing of all Placards and signage to be in compliance with CS/FAR 25.791 Weight, Centre Gravity and Weight Distribution to be in compliance with CS/FAR 25.1519.2. Compliance with CS/FAR 25.786 (g) (3), the elimination of any injurious object within striking radius of the head.3. Compliance with...</td>
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<td>BOEING 737-500</td>
<td>Installation of Galley 3.</td>
<td>AIM Aviation (Henshalls) Definition Document DD 1010 Issuelfew, dated April 11, 2005 or later EASA approved revisions.</td>
<td>1. Viewing of all Placards and signage to be in compliance with CS/FAR 25.791 Weight, Centre Gravity and Weight Distribution to be in Compliance with CS/FAR 25.1519.2. Compliance with CS/FAR 25.785 (d) (3), the elimination of any injurious object within striking radius of the head 3. Compliance with...</td>
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<td>BOEING 737-500</td>
<td>Installation of Wardrobe W1.</td>
<td>AIM Aviation (Henshalls) Definition Document DD 1015 Issuelfew, dated April 25, 2005 or later EASA approved revisions.</td>
<td>1. Viewing of all Placards and signage to be in compliance with CS/FAR 25.791 Weight, Centre Gravity and Weight Distribution to be in Compliance with CS/FAR 25.1519.2. Compliance with CS/FAR 25.785 (d) (3), the elimination of any injurious object within striking radius of the head 3. Compliance with...</td>
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<td>REV 1</td>
<td>AIRBUS A321-200</td>
<td>A321 Livingston - Cabin Modification from 196 Y/C to 210 Y/C with relocation of two attendant seats Note: This revision affects the transfer of this STC from EADS Sogerma Services to Sogerma Services Model: Airbus A321-200 MSN 1950 &amp; 1970</td>
<td>Modification Approval Sheet (MAS) Ref. S/05.25-1102 Issue 01, dated May 16, 2005</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>DASSAULT FALCON SERVICE</td>
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<td>FALCON 10</td>
<td>Antenna installation</td>
<td>Associated to: see Change approval sheet DFS 1263 associated list of affected technical documents</td>
<td>This STC is approved only for the configuration as defined in Mod Sheet DFS 1263.2. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.A.S.01304</td>
<td>EADS SOGERMA</td>
<td>REV 1</td>
<td>AIRBUS A330-200</td>
<td>Installation of a Cabin modification from 248/C - 249/C to 188/C - 271/Y/C with removal of galley G2 and installation of new partitions for LAUDA AIR fleet. The associated modification Approval sheet is referenced S/05.25-1122 issue1 dated 10/06/05. All documents related to this STC are listed in the...</td>
<td>Technical reissue sheet reference 25-1122.</td>
<td>04/01/2008</td>
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<td>EASA.A.S.01305</td>
<td>DASSAULT FALCON SERVICE</td>
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<td>MYSTERE FALCON 50</td>
<td>Introduction of Pressure Seals for wire runs</td>
<td>-see Change approval sheet DFS 5810 and associated list of affected technical documents</td>
<td>This STC is approved only for the configuration as defined in Mod Sheet DFS 5810. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>22/09/2005</td>
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<td>EASA.A.S.01306</td>
<td>UNAIR</td>
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<td>BEECH C90A</td>
<td>Compliance with the B-RNAV requirements for the GPS System Trimble 2000 Approach +</td>
<td>Master Data List UAE C90.05 Revision 2 dated 21 June 2005.AFMS No. 571 UAE Revision 2 dated 21 June 2005</td>
<td>1. This STC is limited to aircraft s/n LJ-1203.2. GPS System Trimble 2000 Approach + installation already approved with IFR privilege.</td>
<td>20/07/2005</td>
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<td>Limitations/Conditions</td>
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<td>BRITISH AIRWAYS PLOBBRITISH AIRWAYS ENGINEERINGOF AIRWORTHINESS</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BOEING 747-400</td>
<td>Omnisus Modification for B747 Mid J Project Stretch Embedment</td>
<td>British Airways Engineering Modification ref. BA.747-25-0007British Airways Engineering Master Data List ref. MDL-0049-06 Rev 00 or later approved revision.</td>
<td>28/02/2007</td>
<td>Active</td>
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<td>EASA.A.S.01315</td>
<td>BRITISH AIRWAYS PLOBBRITISH AIRWAYS ENGINEERINGOF AIRWORTHINESS</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BOEING 777-200</td>
<td>Omnibus Modification for B777 4 Class Project Stretch Embedment.</td>
<td>British Airways Master Data List Reference No. MDL-0055-08 Rev 00 dated 13 June 2008 or later approved revision. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>09/09/2008</td>
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<td>EASA.A.S.01316</td>
<td>TMC AERO</td>
<td>AIRBUS A330-243</td>
<td>This STC deals with the installation of a TMC stretcher. All documents related to this STC are listed in the STC form A330-25-23. The associated &quot;Fiche de Modification&quot; from Corsair is referenced 25-137 Rev. 1 ( or later EASA-approved revision</td>
<td>The approval of this modification is limited to the Airbus A330-243. The limitations and conditions defined in the EASA Datasheet are applicable.</td>
<td>02/06/2005</td>
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<td>10014179</td>
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<tr>
<td>EASA.A.S.01318</td>
<td>TMC AERO</td>
<td>BOEING 747</td>
<td>The installation of stretchers mounted on the seat tracks in passenger cabin of the Boeing 747 for the transport of patients, as described in Corsair Fiche de Modification no. 25-136</td>
<td>- Corsair Note Technique DT:ET/04129NT; Corsair Fiche de Modification no. 25-136; JTSO F.O. 103 for model TMC 99</td>
<td>1. The maximum number of passengers allowed when two stretchers are installed shall be determined.2. When three or more stretchers are installed transport of passengers is not allowed.</td>
<td>12/07/2005</td>
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<td>10014181</td>
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<td>EASA.A.S.01320</td>
<td>ATS INTERNATIONAL</td>
<td>B757</td>
<td>Installation of emergency Locator Transmitter</td>
<td>ATS International Modification STC 0248-01ATS InternationalMaster Data List 0248-01A-2103-R00</td>
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<td>16/10/2006</td>
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<td>EASA.A.S.01323</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>FOKKER F.27 MARK 600 FOKKER F.27 MARK 500</td>
<td>Use of TRA-46A Display Indicator</td>
<td>- STC Twenty One Modification S21.34-62-1013, STC Twenty One AFM Supplement AEDF2718 Issue 2 or later approved revision</td>
<td>30/01/2006</td>
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<td>EASA.A.S.01324</td>
<td>AERO ENGINEERING DESIGN LTD</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BAE 125 800B</td>
<td>Introduction of Rockwell Collins TCAS 4000</td>
<td>Flight Manual Supplement S.21 FMS.125.0004</td>
<td>19/09/2005</td>
<td>Active</td>
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<td>EASA.A.S.0133</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA7076NM</td>
<td>BAE 14 146 AND AVRO 146 MODELS</td>
<td>Installation of Stretcher Carrier in accordance with AIR MEDIC MasterDrawing List Ref: AM-2001, Revision A2, dated May 18, 1995 (or later/EASA approved Revisions).</td>
<td>1 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or rev...</td>
<td>16/09/2004</td>
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<td>EASA.A.S.01331</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>CESSNA 525</td>
<td>Installation of ACAS II System with Dual Mode S Transponder. ELTs compliant and EHS capable</td>
<td>AFM Supplement Doc. No. 118A-AFMS, revision 0 or later/EASA approved revision.</td>
<td>18/07/2005</td>
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<td>EASA.A.S.01332</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>CESSNA 550 CITATION II CESSNA 500, CITATION</td>
<td>Dual GARMIN GPS Installation: GNS-430A + GNS-530A in accordance with MDL BCA 74-02 rev 6 dated 15/07/05</td>
<td>AFMS SMY BCA74 rev 7 dated 11/01/2005 MMY BCA74 rev 0 dated 07/07/05</td>
<td>31/01/2007</td>
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### List of EASA Supplemental Type Certificates

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<th>STC Holder</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC</td>
<td>A330-342/343</td>
<td>Installation of Stretcher Carrier in accordance with AIR MEDIC MasterDrawing List Ref. - AM-2001, Revision JXY, dated April 15, 2002 (or later EASA approved Revisions)</td>
<td></td>
<td></td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/ engine configurations shall be determined by the installer. This STC shall remain valid unless otherwise surrendered or revoked.</td>
<td>16/09/2004</td>
<td>Active</td>
<td>10261303</td>
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<tr>
<td>AIRBUS S.A.S.</td>
<td>REV 1</td>
<td>A330-200 MODELS</td>
<td>Installation of a more powerful air heater (3kg W) inside the private cabin washroom</td>
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<td>04/07/2006</td>
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<td>10041357</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV 1</td>
<td>BEECH B300</td>
<td>Installation of TAWS ST3400 system Revision of the AFM Supplement related to a Minor Change to EASA.A.S. Issue 01340 Installation of TAWS ST3400 system Revision of the AFM Supplement related to a Minor Change to EASA.A.S. Issue 01349.</td>
<td></td>
<td>AFM Supplement No. 2022641-AM-03, Issue 3 or later EASA approved revision. Continued Airworthiness Instruction 2022641-AM-03 EASA approved Aircraft Flight Manual Supplement Doc No. 2022641-AM-03, Issue 4, dated January 2007 or later approved revision.AFM Supplement No. 2022641-AM-03, Issue 3 or...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/ engine configurations shall be determined by the installer. This STC is approved only for the product configuration as defined in...</td>
<td>26/02/2007</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>CESSNA 500</td>
<td>Installation of Honeywell KMH 880 Multi-Hazard Awareness System</td>
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<td>02/01/2006</td>
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<td>E.A.D AEROSPACE AIRWORTHINESS</td>
<td>REV 1</td>
<td>SA 227 SERIES</td>
<td>ELS/EHS installation on SA 227 series aircrafts- Removal of existing two ATC transponders and associated trays in forward avionics- Removal of existing ATC Mode S Control Panel from Instrument Panel- Disconnection of ATC 2 Mode S bottom antenna- Installation of two ATC transponders on new trays in re...</td>
<td></td>
<td>Aircraft Flight Manual Supplement to installation of Honeywell KMH 880 Multi-Hazard Awareness System Doc No. 2027746 AM-01 Issue 1 dated May 2005 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/ engine configurations shall be determined by the installer. This STC is approved only for the product configuration as defined in...</td>
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<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
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<td>A300F4-606ST</td>
<td>Supplement to A300-606ST Flight Manual referred ST34400027 dated 28 March 2008 or any later EASA approved revision</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>Installation of VOL 4000A VHF Data Link Mode 4 (CNS Systems) and installation of a combined VHF/GPS Antenna</td>
<td>Master Document Lot Doc. No. 045A-MLD Rev. N/C or later EASA approved revisions. Continued Airworthiness Plan Doc. No.045A CAWP</td>
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<td>DAO AVIATION/IT</td>
<td>Installation of Garmin GNS 430 COM/NAV/GPS; Approval for B-RNAV, en route IFR and Terminal flights</td>
<td>DFE Design Order Package DO-0011. Garmin installation manual PN: 190-00140-02 (Rev. L) SLV approved Aircraft Flight Manual supplement DFE- DG-AMFS-01-00</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>Introduction of Freight Loading &amp; Restraint System for LFD Modified Aircraft.</td>
<td>STC 21 Master Drawing List No. S21-MLD-0018 initial issue dated 25 January 2008 or later approved revision (STC 21 Airplane Flight Manual Supplement No. S21- FMS-0009 at initial issue or later approved revision.</td>
<td>This major change is limited to installation of a large cargo door at the rear of the fuselage in accordance with BAe Service Bulletin ATP-52-036 and with the installation of a main cabin Class 'E' cargo compartment in accordance with STC Twenty One Limited Master Document List S21-MLD-0017.</td>
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<td>09/04/2008</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>Installation of AAR Freight Loading System</td>
<td>-STC21 Change No: S21-29-10-04S STC21 AFM Supplement Serial No S21.FMS-ap-0010 Issue No. or later approved revision</td>
<td>This STC may only be embodied into BAE ATP aircraft previously modified for the installation of a large cargo door at the rear of the fuselage in accordance with British Aerospace Service Bulletin SBATP-52-038 (Modification MI0139A), with the barrier net installed in accordance with FAA Advisory Circular 21-101.</td>
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<td>30/01/2006</td>
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<td>AVIONIK STRAUBING ENTWICKLUNG GmbH</td>
<td>Installation of Dual Garmin GNS 430A</td>
<td>EASA approved AFM Supplement EASA2005-032-00 Rev. A, or later EASA approved issues</td>
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<td>EASA.A.S.01378</td>
<td>DASSAULT FALCON SERVICE</td>
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<td>MYSTERE FALCON 900</td>
<td>Call out evolution with EGPHS Honeywell Mark V 212-212. This STC superseded the STC F-C163-SF-2107 EASA approved on March 9, 2004 with the reference 2004-2073.</td>
<td>Change Approval Sheet DFS 9447 CES revision A dated June 9, 2005 or later approved - AFM supplement DFS no 17 revision A dated June 9, 2005.</td>
<td>Applicable to all MF900 aircraft with modification DFS 9259</td>
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<td>04/07/2005</td>
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<td>EASA.A.S.01384</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
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<td>BAE.125 SERIES 800B BAE.125 SERIES 800A</td>
<td>Installation of a Rockwell Collins TCAS 4000 with Dual TDR-94D and Dual RTU-4210 System</td>
<td>Engineering Order Doc. No. 206-HW Rev. IR or later EASA approved Revisions/Aircraft Flight Manual Supplement/Collins TCAS4000 with Dual TDR-94D and Dual RTU-4210, Doc. No. AG530-071 Rev IR or later EASA approved - Revisions/Aircraft Flight Manual Supplement/Collins TCAS4000 with Dual TDR-94D and Dual RTU-4210, Doc. No. AG530-071 Rev IR or later EASA approved - Revisions/Aircraft Flight Manual Supplement/Collins TCAS4000 with Dual TDR-94D and Dual RTU-4210, Doc. No. AG530-071 Rev IR or later EASA approved</td>
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<td>10/11/2005</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.01388</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>AIRBUS A321 AIRBUS A320</td>
<td>ATC, System Upgrade for Enhanced Surveillance, acc.: STD 34-05/06M</td>
<td>Aerospace</td>
<td>AFM Supplement A-34-05/260-AMICA-DR04-0379-02</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>none</td>
<td>22/07/2005</td>
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<td>EASA.A.S.01390</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>AIRBUS A330-300</td>
<td>Integration of Lower Deck Mobile Crew Rest</td>
<td>Aerospace</td>
<td>Supplemental Type Definition Document: STD00-04/11M Issue 01/Compliance Record List: CRL00-04/11 Issue 1Instructions for Continued Airworthiness</td>
<td>Emergency evacuation procedures, including the emergency evacuation of an incapacitated occupant from the Crew Rest Area, must be established. All of these procedures must be transmitted to the operator for incorporation into their training programs and appropriate operational manuals.</td>
<td>none</td>
<td>14/07/2005</td>
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<td>EASA.A.S.01391</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>AIRBUS A340-300</td>
<td>Integration of Lower Deck Mobile Crew Rest</td>
<td>Aerospace</td>
<td>Supplemental Type Definition Document: STD00-04/11M Issue 01/Compliance Record List: CRL00-04/11 Issue 1Instructions for Continued Airworthiness</td>
<td>Emergency evacuation procedures, including the emergency evacuation of an incapacitated occupant from the Crew Rest Area, must be established. All of these procedures must be transmitted to the operator for incorporation into their training programs and appropriate operational manuals.</td>
<td>none</td>
<td>14/07/2005</td>
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<td>EASA.A.S.01392</td>
<td>FINNAIR OYJ</td>
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<td>BOEING MD 11</td>
<td>MD11 Cabin Crew Rest Area</td>
<td>Aerospace</td>
<td>Data package containing the following documents, and additional supporting data: 1. MD-11 C/A Bunk Airworthiness Compliance Sheet Rev. E2: Cabin Crew Rest Compartment Modifications 3. MD-11 C/A Bunk AYM60 and AYM61 Flight Test Report Rev. 04: MD-11 C/A Bunk AYM60 and AYM61 Ground Test Rev.</td>
<td>Limited to S/N 2843066: F-GSMDGPS use is forbidden for BRNAV and approach Auto Pilot coupling is forbidden under 2000 ft AGL</td>
<td>none</td>
<td>15/08/2006</td>
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<td>EASA.A.S.01393</td>
<td>RECTIMO AVIATION</td>
<td></td>
<td>PIPER PA-28-140, ARCHER III</td>
<td>GNS 430 Installation with MD-41-1488 (HSI and AP coupled) Installation of GNS 430 and MD-41-1488 coupled with KI 525A</td>
<td>Aerospace</td>
<td>FACE RECTIMO AVIATION 2005/05-0 FSIM4D/4MS &quot;Supplement au manuel de vol apprové EASA concernant l'utilisation de l'avitlonéquipé dans GPS CARMIN GNS 430; Edition juillet 2007”</td>
<td>Limited to S/N 2843066 F-GSMDGPS use is forbidden for BRNAV and approach Auto Pilot coupling is forbidden under 2000 ft AGL</td>
<td>none</td>
<td>20/07/2007</td>
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<td>EASA.A.S.01394</td>
<td>AIRBUS S.A.S.</td>
<td></td>
<td>A340-212</td>
<td>VIP Cabin and systems upgrade in accordance with Sogerma Modification Approval Sheet (MAS) SOG-00-0377 Issue 3 dated 02/02/08</td>
<td>Aerospace</td>
<td>Flight Manual Supplement ref: FM-S MOD-EOG F4-00-977 issue 6 dated 01/02/2008 MPD Supplement ref: MPO-S SOG-F4-00-977 dated 31/01/2008 or later approved revision</td>
<td>The approval of this modification is limited to the Airbus A340-212 MSN 061</td>
<td>none</td>
<td>04/07/2008</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.01396</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>A319-133</td>
<td>Airbus Rack 9100 VU installation</td>
<td>― Modification Approval Sheet (MAS) ref. SGG-00-1056, issued dated October 4, 2005- AFM Supplement, ref. FM- S MOD-SOG-00-1056, dated September 21, 2005- MMEL Supplement, ref. MMEL-S MOD- SOG-00-1056, dated September 21, 2005</td>
<td>1. This STC is limited to A319-133 CJ MSN 2421</td>
<td>04/07/2008</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.01397</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
<td></td>
<td>A319-133</td>
<td>Live TV system provision</td>
<td>Modification Approval Sheet (MAS) ref. SGG-00-1159, issue 1 dated October 4, 2005</td>
<td>This STC is limited to A319-133 CJ MSN 2421</td>
<td>04/07/2008</td>
<td>Active</td>
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<td>EASA.A.S.01399</td>
<td>UNI AIR ENTREPRISE</td>
<td></td>
<td>CORVETTE SN 601 SN 028 &amp; 008</td>
<td>Deviation to SB 34-12 replacement of mount C24-C25 in accordance with MDL UAE 008.S Rev. 1 dated 8/07/2005. Installation of horizontal mount between frames C24 and C25 instead of vertical one between frames C24 and C25.</td>
<td>AFMS No. 58</td>
<td>1. This STC applies to MSN 028 and MSN 008 only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>25/07/2005</td>
<td>Active</td>
<td>10014236</td>
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<tr>
<td>EASA.A.S.0140</td>
<td>AIR MEDIC L.L.C.</td>
<td>FFA STC, SAA490NM</td>
<td>A300 / A310 SERIES</td>
<td>Installation of Stretcher Carrier in accordance with AIR MEDIC MasterDrawing List Ref. AM-2001, Revision &quot;o&quot;, dated March 11, 1987 (or later/EASA approved Revisions)</td>
<td></td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>16/09/2004</td>
<td>Active</td>
<td>10014237</td>
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<td>EASA.A.S.01400</td>
<td>UNAIR</td>
<td>CORVETTE SN 601</td>
<td>Deviation to SB 34-12 - replacement of ADF in accordance with MDL UAE 019.05- rev 1 dated 11/07/2005. Installation of ADF according to option 85 aeroisolate instead of ADF 206. Installation of option 85 on second ADF position instead of first position. plane model: Airbus Corvette SN 801, MSN 028 and 050</td>
<td></td>
<td>1. This STC is limited to S/N 028 and S/N 082. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>28/07/2005</td>
<td>Active</td>
<td>10014238</td>
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<td>EASA.A.S.01403</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>CESSNA F182-Q</td>
<td>Installation of propeller MTV-9-Q203-52</td>
<td>Anhang zum Flughandbuch Nr. E-1243/Airplane Flight Manual Supplement No. E-1244/Instructions for Continued Airworthiness and Installation Instructions E-124</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>18/11/2005</td>
<td>Active</td>
<td>10014239</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.01452</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>MYSTERE FALCON 20</td>
<td>Deviation to M GE 2 - Pressure seals FR17-18</td>
<td>see Change approval sheet DFS 2651 and associated list of affected technical documents</td>
<td>1. This STC is applicable to Falcon 20/S/N 103 only. 2. This STC is to be applied simultaneously with Dassault Aviation Mod M GE23. Compatibility with other aircraft/ engine configurations shall be determined by the installer</td>
<td></td>
<td></td>
<td>22/08/2005</td>
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<tr>
<td>EASA.A.S.01456</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>BOEING 767-300</td>
<td>ATP, System Upgrade for EHS</td>
<td>Supplemental Type Definition Document STD034-045M Issue 01, Rev 01 or later approved revision Airplane Flight Manual Supplement A-34-04670-00-AP Rev IR or later approved revision</td>
<td>Certification Basis of the Change: CS 25 Change 0, additionally JAA Leaflet No 18 (Mode-S EHS)</td>
<td></td>
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<td>10/04/2006</td>
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<tr>
<td>EASA.A.S.01471</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td></td>
<td>LEARJET 55C LEARJET 55</td>
<td>Installation of Flight Data Recorder F1000</td>
<td>- Engineering Order 2003-8, Rev B or later EASA approved revisions</td>
<td></td>
<td></td>
<td></td>
<td>01/08/2005</td>
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<tr>
<td>EASA.A.S.01473</td>
<td>ELECTRONIQUE MAINTENANCE AVIATION</td>
<td></td>
<td>BEECH 200</td>
<td>406 MHz ELT installation</td>
<td>Airplane Flight Manual supplement No. EMATA/002E000/ EL743E, rev. 1 of July 2005.</td>
<td>Modification applicable on PN B.B8619 and B.B8620 only.</td>
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<td></td>
<td>05/06/2005</td>
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<tr>
<td>EASA.A.S.01477</td>
<td>STC TWENTY ONE LIMITED</td>
<td></td>
<td></td>
<td>Introduction of Honeywell MKV Enhanced Ground Proximity Warning System (EGPWS)</td>
<td>Master Data List S21.MDL-0001-Issue 2 or later approved revision AFM Supplement S21.FMS-0001 Issue 1 or later approved revision</td>
<td></td>
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<td>22/09/2005</td>
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<tr>
<td>EASA.A.S.01478</td>
<td>STC TWENTY ONE LIMITED</td>
<td></td>
<td></td>
<td>Introduction of Honeywell GNSSU Receiver</td>
<td>Master Data List S21.MDL-0002-Issue 1 or later approved revision.</td>
<td></td>
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<td>22/09/2005</td>
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<td>EASA STC no.</td>
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<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>EASA.A.S.01481</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>MYSTERE FALCON 20 FAN JET</td>
<td></td>
<td>DFS 2545 dtd. 12 October 05: Camera and Video Recorder installation</td>
<td>see Change approval sheet DFS 2545 and associated list of affected technical documents</td>
<td>This STC approves the installation only, system activation is not considered.</td>
<td>08/11/2005</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.01486</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV 1</td>
<td>AIRBUS A330-300 AIRBUS A330-200</td>
<td></td>
<td>Structural mounting provision for large antenna and random for Live TV Tailwind 560</td>
<td>Airplane Flight Manual Supplement: A-00-06/138- AFM Rev. II/0 later approved revision.Supplement Type Definition Document Ref. STD0 53-05/02M, Issue 02 / Rev. 00 dated 7 December 2006</td>
<td>1. EASA CS25, without Amendment, Certifications Specifications for Large Aircrafts plus JAA Generic Issue Paper A-X &quot;Structural Certification Criteria for Large Aircrafts &amp; Systems&quot;. This STC is approved only for the product configuration as defined in the approved design data referred to in the...</td>
<td>08/12/2006</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.01488</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV 1</td>
<td>BOEING 777-200</td>
<td></td>
<td>Rearrangement of VIP cabin interior (Majlis) &amp; systems and modification of pax entertainment system (IFE) including installation of Tailwind 560 Live TV system.</td>
<td>STC Definition Document - STD00-05/06M issue 01 dated 11 June 2006 (or later approved issue) Compliance Record List CR/06-05/06M issue 10.Instructions for Continued Airworthiness - A-00-05/42-ICA Rev IR (or later approved issue/EASA CR/A-01 is 4 Certification Basis VIP Interior.</td>
<td>1. This modification is limited to a/c serial number MSN29271.2. Occupancy of this zone is limited to Zero Passengers for Taxi, Take-off and Landing, pending substantiation of evacuation through the adjacent compartments.</td>
<td>12/06/2008</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.01489</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>BOEING 777-200</td>
<td></td>
<td>Introduction of Tailwind 560 Antenna and Live TV for VP 8777-200.</td>
<td>STD003-05/01M issue 01 dated 22 June 2006AFM Supplement: A-00-05/342-AFM/Instructions for Continued Airworthiness CS53-0223-2</td>
<td>This modification is limited to a/c serial number MSN29271.</td>
<td>29/06/2006</td>
<td>Active</td>
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<td>EASA.A.S.01490</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>AIRBUS A300</td>
<td></td>
<td>ATC, System Upgrade for EHS</td>
<td>Supplemental Type Definition Document STD004-05/07M, Issue 1, Rev. 00 or later approved Revision.Airplane Flight Manual Supplement: A-34-05/323-AFM Rev. IR, dated Jul 2005 or later approved Revision</td>
<td>Certification Basis of the Change: CS 25, additionally JAA Leaflet No. 18 (Mod-S EHS)</td>
<td>13/03/2006</td>
<td>Active</td>
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<td>EASA.A.S.01493</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>FALCON 20</td>
<td></td>
<td>Installation of two V/UHF AN/ARC210</td>
<td>Associated to: see Change approval sheet DFS 2543 and associated list of affected technical documents</td>
<td>1. This STC is applicable to Falcon 20 SN 076 only. System activation is not considered.3. This STC is approved only for the configuration as defined in ModSheet DFS 2543.4.Case of other aircraft/ engine configurations shall be determined. This STC is approved only for the product configuration...</td>
<td>09/08/2005</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.01494</td>
<td>AIR FRANCE</td>
<td></td>
<td>B767</td>
<td></td>
<td>Rockwell Collins Aio I SATCOM installation</td>
<td>CAS B767-05-004 Rev 0</td>
<td></td>
<td>10142194</td>
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<td>EASA.A.S.01496</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>MYSTERE FALCON 20</td>
<td></td>
<td>Antenna relocations, skin repairs</td>
<td>see Change approval sheet DFS 2558 and associated list of affected technical documents</td>
<td>1. This STC is only approved for the configuration as defined in ModSheet DFS 2558 Issue Original</td>
<td>14/09/2005</td>
<td>Active</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>EASA.A.S.01498</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>BOEING 737-700 IGW</td>
<td>Cabin and IFE Modification as defined in Supplemental Type Definition Document STD00 - 05/05 M, Issue 01, Rev.00</td>
<td>Lufthansa Technik Supplemental Type Definition Doc. No. STD00 - 05/05 M, Issue 01 Rev. 00 or later approved revision</td>
<td>1. Applicable to Falcon 20 S/N 103 only.</td>
<td>25/10/2005 Active 10014269</td>
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<td>EASA.A.S.01500</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>FALCON 10</td>
<td>GPS Antenna installation</td>
<td>Associated to: see Change approval sheet DFS 2563 and associated list of affected technical documents</td>
<td>1. This STC is limited to MF 20 GF SN 397.2. Dassault Aviation M2719 to be applied (hard points installation). Only symmetrical configurations are allowed (1+1 or 2+2). Maximum Landing Weight is limited to 11.816kg. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>16/08/2005 Active 10014270</td>
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<td>EASA.A.S.01502</td>
<td>CESSNA DUESSELDORF CITATION SERVICE CENTER GmbH</td>
<td></td>
<td>CESSNA 551</td>
<td>Installation of Honeywell RDS 84 WX System.</td>
<td>MM-Supplement 2004-08-10, Rev. A or later approved revision.</td>
<td>1. This STC is applicable to Falcon 20 SN 076 only. System activation is not considered. 2. This STC is approved only for the configuration as defined in Mod Sheet DFS 2548. 4. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>16/08/2005 Active 10014272</td>
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<td>EASA.A.S.01506</td>
<td>DASSAULT FALCON SERVICE</td>
<td>REV. 1</td>
<td>MYSTERE-FALCON 20 SN 397 FAN JET</td>
<td>see Change approval sheet DFS 2483A-ARM Supfo MF 20 GF 47 Revision 1 Utilisation de l'avion equipé de quatre sondes PMISous volet and associated list of affected technical documents</td>
<td>1. This STC is limited to MF 20 GF SN 397.2. Dassault Aviation M2719 to be applied (hard points installation). Only symmetrical configurations are allowed (1+1 or 2+2). Maximum Landing Weight is limited to 11.816kg. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>15/05/2006 Active 10043014</td>
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<td>EASA.A.S.01507</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td></td>
<td>B90</td>
<td>Installation of Avidyne FM850</td>
<td>-EASA approved AFM Supplement LBA2004-015-00 Rev. A or later EASA approved issues</td>
<td>1. This STC is applicable to Falcon 20 SN 076 only. System activation is not considered. 2. This STC is approved only for the configuration as defined in Mod Sheet DFS 2548. 4. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>18/08/2005 Active 10014273</td>
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<td>EASA.A.S.01510</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>FALCON 20</td>
<td>Time Base Installation</td>
<td>Associated to: see Change approval sheet DFS 2548 and associated list of affected technical documents</td>
<td>1. This STC is applicable to Falcon 20 SN 076 only. System activation is not considered. 2. This STC is approved only for the configuration as defined in Mod Sheet DFS 2548. 4. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>18/08/2005 Active 10014274</td>
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<td>EASA.A.S.01511</td>
<td>SERAM - SOCIETE D'ENTRETIEN ET REPARATION AERONAUTIQUE MACONNAISE</td>
<td></td>
<td>CESSNA 337D</td>
<td>Installation of one KK 155A Immunity FM, one ONS430 and G106A with BRNAV capability</td>
<td>Flight Manual Supplement Rev 0 dated 01/09/2004</td>
<td>1. This STC is limited to Cessna 337D SN 337-1046. 2. Autopilot coupled with ONS 430 forbidden. 3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>23/08/2005 Active 10014275</td>
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<td>EASA.A.S.01512</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>FALCON 10</td>
<td>Frame 0 Pressure Seal through-fitting modification (enlarge pressure seals)</td>
<td>Associated to: see change approval sheet DFS 1272</td>
<td>This STC is approved only for the configuration as defined in Mod Sheet DFS 1272 Issue Original. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>24/08/2005 Active 10014276</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd</td>
<td>REV 2</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BAE 146/RJ SERIES</td>
<td>Installation of PRNAV Compliant UMS-1E FMS</td>
<td>- Bouremouth Aviation (Consultants) Modification BAC/146064 Issue 1 or later approved revision. - Bouremouth Aviation (Consultants) Modification BAC/146066 - Bouremouth Aviation (Consultants) AFM Supplement FMS/0490 issue 2 or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>08/12/2006</td>
<td>Active</td>
<td>1014277</td>
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<td>AEROTECH &amp; CONCEPT</td>
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<td>PIPER PA-31</td>
<td>Installation of Radio Navigation and Communication Systems-IX 175B, KY 196, KN 53, KE 75, EMA 24 and KN 65 removal- GNS 430, XX 155A, MD 41, RDR 2050, GMA 340, KN 62A, KI 229, GI102A installation</td>
<td></td>
<td>The approval is limited to MSN 31-367 F-GDPU</td>
<td>20/02/2006</td>
<td>Active</td>
<td>1014278</td>
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<tr>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>2</td>
<td>30, 30, 300LV 1, B200, B300</td>
<td>200, 200C, 300, 300LV 1, B200, B200C, B300</td>
<td>Installation of Dual Mode-S Transponder System Enhanced Surveillance (Exchange of existing transponder)</td>
<td>EASA Approved Airplane Flight Manual Supplement B300-123667-06/01, Revision 1, dated 20.05.2006 or later EASA approved revision.</td>
<td></td>
<td>13/06/2006</td>
<td>Active</td>
<td>1014279</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td></td>
<td>BOEING 757-300</td>
<td>Installation of Zonal Drying System</td>
<td>Supplemental Type Definition Document STD201-05/02M, Issue 1, Rev. 0, or later approved revision</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this previously installed modification and any other installation of this or any other modification the installer must determine that the interrelationship between this previously installed modification and any other installation of this or any other modification the installer must determine that the interrelationship between this previously installed modification and any other installation of this or any other</td>
<td>06/11/2007</td>
<td>Active</td>
<td>1014280</td>
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<tr>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td></td>
<td>MYSTERE FALCON 20</td>
<td>Power Utility</td>
<td>see Change approval sheet DFS 2649 associated list of affected technical documents</td>
<td>System activation is not considered</td>
<td>28/09/2005</td>
<td>Active</td>
<td>1014294</td>
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<tr>
<td>AERODISA S.L.</td>
<td></td>
<td></td>
<td>DC-8-62</td>
<td>Revision to the cargo modification on DC-8-62</td>
<td>Documentation referred in Doc. AED02000-01M Issue 01, April 2006 Maintenance Manual Supplement AED02000-01M, SMM Issue 01, April 2006.</td>
<td>Limited to DC-8-62 series airplanes which have been converted from a passenger to a cargo-carrying (&quot;Freighter&quot;) configuration by FAA STC No. SA185250, but with a deviation in therein deck cargo 9g crash barrier that was installed following FAA STC No. SA137750.</td>
<td>18/04/2007</td>
<td>Active</td>
<td>1014296</td>
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<td>EASA STC no.</td>
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<td>EASA.A.S.01535</td>
<td>LUFTHANSA TECHNIC AG</td>
<td>REV 1</td>
<td>BOEING 747-SP</td>
<td>ATC System, Upgrade to Enhanced Surveillance</td>
<td>Supplemental Type Definition Document No. STD204-05/08, M, Issue 2, Rev. 05, dated 15 January 2007 or later approved revision; Airplane Flight Manual Supplement A-34-06/49/AFM, Revision IP, dated 22 Nov 2006 or later approved revision</td>
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<td>1. Certification Basis of the change: CS 25 Change O, additionally JAA Leaflet No. 18 (Mode-S EHS)2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/ engine configurations...</td>
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<td>EASA.A.S.01536</td>
<td>DASSAULT FALCON SERVICE</td>
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<td>MYSTERE FALCON 20</td>
<td>DFS 2562 dist 21 Dec 05: Master Mission Function</td>
<td>see Change approval sheet DFS 2562 DFS AFM Supplement F2009 and associated list of affected technical documents</td>
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<td>1. This STC is approved only for the configuration as defined in Mod Sheet DFS 2562. Compatibility with other aircraft/ engine configurations shall be determined by the installer</td>
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<td>EASA.A.S.01545</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>C90</td>
<td>Installation of radio communication systems Model: Beechcraft C90 SN LJ 710</td>
<td>Supplemental AFM for GNS 330 edition 1 of 15 January 2005, revision 2 Supplemental AFM for GNS 430 edition 1 of 15 January 2005, revision 2 and Supplemental AFM for KMD 850 edition 1 of 15 January 2005, revision 2</td>
<td>1. This STC is limited to Beechcraft C90 SN LJ 710. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>12/09/2005</td>
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<td>EASA.A.S.01548</td>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td>BEECH S35</td>
<td>Installation of following equipment: KN63+KDS72+KH165+K203+HMD150+ AR4401+WX400+ANT NY163</td>
<td>FAM No 299/03 and AFMS for KMD 150 No 2 - Revision 0</td>
<td>This STC is applicable only for the Beechcraft S35 - S/N D7498</td>
<td>16/09/2005</td>
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<td>EASA.A.S.01550</td>
<td>AERLYPER S.A.</td>
<td>BEECH 200</td>
<td>Installation of MST67A Transponder Mode S Equipment</td>
<td>Master Data List AERLYPER DOC N° 100469 Ed. 04 Instruction for continuous Airworthiness AERLYPER DOC: N° 100482 Ed. 01</td>
<td>06/12/2007</td>
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<td>EASA.A.S.01551</td>
<td>SOGERMA SERVICES</td>
<td>ATR 42-320</td>
<td>Additional probes in RALI configuration in accordance with MAS SQG-01-459 issue 1 dated 30/11/2004. This revision reflects the transfer of this STC from EADS Sogema Services to Sogema Services.</td>
<td>AFM supplement Ref 2400/45 SQG/DEV issue 4</td>
<td>1. MSN 078 Associated STC No.EASA.A.S.01219 Revision 1</td>
<td>04/01/2008</td>
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<td>EASA.A.S.01553</td>
<td>LUFTHANSA TECHN K AG</td>
<td>AIRBUS A319</td>
<td>Installation of new B/E Aerospace convertible and standard seats and cabin refurbishment to PG-design on Lufthansa’s A319 aircrafts</td>
<td>Certification Program CPS-05-02M, issue 03, dated 18 Jan 2006, Supplemental Type Definition Document STD005-02M, issue 01 dated 30/01/2006, Compliance Record List CRL05-052M, issue 01 dated 24/01/2006, Applicant’s Declaration of Compliance dated 03/02</td>
<td>16/02/2006</td>
<td>Active 1001435</td>
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<td>EASA.A.S.01555</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td>CAE UK RESPONSIBILITY</td>
<td>BAE 146</td>
<td>Change EGPS5 Altitude Callouts</td>
<td>&quot;Bournemouth Aviation (Consultants) Ltd Modification BAC146/057 Issue 1 or later approved revision&quot;</td>
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<td>16/03/2006</td>
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<td>EASA.A.S.01556</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>525</td>
<td>Installation of SKY497 and TAWS</td>
<td>800! The SKY497 is wired to the new Antenna, existing AM-250 Copilot Audio. The TAWS 800 is wired to the existing MFD, KLJN90B and ADC 200</td>
<td>EASA approved AFM Supplement FMS A09 2005-023-00 Rev. A, or later approved revisions</td>
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<td>EASA.A.S.01558</td>
<td>FR AVIATION Ltd/DIBA COBHAM AVIATION SERVICES UK</td>
<td>CAA UK RESPONSIBILITY</td>
<td>FALCON 20 BASIC D AND E SERIES</td>
<td>DC-0038: ACAS II and Mode S Installation</td>
<td>see FRA Certification Document CP-0038/AFM Supplement FRAH/FA/008 to AFM UK Supplement DTM918</td>
<td>1. This STC is approved only for the configuration as defined in DC-0038. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft or configurations shall be determined by the installer.</td>
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<td>EASA.A.S.01567</td>
<td>AERLYPER S. A.</td>
<td>AEROSPATIALE ATR 42-300 SERIES</td>
<td>Calibration</td>
<td>Installation of a new antenna with EAS-8000The SKY497 is wired to the new Antenna, existing AM-250 Copilot Audio. The TAWS 800 is wired to the existing MFD, KLJN90B and ADC 200</td>
<td>Maintenance Planning Document Supplement ref. 01050-EAS-A-MPD-01.</td>
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<td>EASA.A.S.01568</td>
<td>AERLYPER S. A.</td>
<td>BEECH 200, SUPER KING AIR</td>
<td>Immaterial SATCOM installation on board B200. The modifications optional and fixed. It consists of an Immaterial SATCOM terminal to be installed in the passenger cabin. A SATCOMAntenna is installed on the upper part of the fuselage.</td>
<td>Documentation referred to in Doc. No. 100528 &quot;Aircraft Flight Manual Supplement&quot;, Issue 1, 27 November 2005.</td>
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<td>EASA.A.S.01569</td>
<td>AERLYPER S. A.</td>
<td>BEECH 200</td>
<td>Installation of Airborne Equipment to comply with COMNAV regulations for IFR and B-RNAV Operation.</td>
<td>Master Data List AERLYPER DOC N° 100564 Ed. 415 Airworthiness Supplement, Aircraft Flight Manual AERLYPER DOC N° 100565 Ed. 415 Instructions for Continuous Airworthiness AERLYPER DOC N° 100566 Ed. 01</td>
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<td>EASA.A.S.01573</td>
<td>BBJ DESIGN SERVICES LIMITED</td>
<td>CAUK RESPONSIBILITIES</td>
<td>A330</td>
<td>Installation of In Seat Power Supply Korean Air Airbus A330</td>
<td>BBJ Design Services Modification BBJ569</td>
<td>DFE Design order package DO-0059 Honeywell installation manualAntenna installation design by Raytheon Aircraft Company Ref: no. JBB/0416-08 Drawing kit P/N 101-4092, rev 1.</td>
<td>MSN BB-309 only.</td>
<td>25/01/2006</td>
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<td>EASA.A.S.01577</td>
<td>DAO AVIATION A/S</td>
<td>BEECH 200, SUPER KING AIR</td>
<td>STC ref.</td>
<td>Installation of Honeywell ACAS II system comprising CAS 67AProcessor, 2 IVA-81D VSIs and 2 ANT-67A antennas.</td>
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<td>EASA.A.S.01579</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>B737-800</td>
<td>Business Class Installation with Sicma Seats</td>
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<td>JET AVIATION VIENNA GMBH</td>
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<td>S/N 560-0260 AND ON</td>
<td>560</td>
<td>Upgrade of the existing MK V1 Ground Proximity Warning System to MK VIII (Enhanced Ground Proximity Warning System)Cessna 560 Citation Ultra, S/N 560-0260 and on, equipped with Honeywell Integrated Avionics System, Honeywell Primus O II SRZ Integrated Radio System and MK VI GPWS</td>
<td>ARS Engineering Order 5426/05/2</td>
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<td>EASA.A.S.01581</td>
<td>JET AVIATION VIENNA GMBH</td>
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<td>560</td>
<td>Integration of a Traffic Alert and Collision Avoidance System and update of the ATC Mode S Transponders to Elementary Mode S Cessna 560 Citation Ultra, S/N 560-0260 and on, equipped with Honeywell Integrated Avionics System, Honeywell Primus O II SRZ Integrated Radio System</td>
<td>ARS Engineering Order 5426/05/1</td>
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<td>EASA.A.S.01582</td>
<td>CESSNA DUESSELDORF CATIONSERVIC E CENTER GmbH</td>
<td>CESSNA 560 (CATION V)</td>
<td>S/N 560-0260 AND ON</td>
<td>560</td>
<td>Installation, Completion of custom Aircraft Interior for B7D00-1A14 Global Express...</td>
<td>AFM Supplement G- FM2504001 Rev N/C or later TCCA approved rev.ICA G-9450000- IPCAAAM Supplement G- TMS0104001 Ref: G-FM2504001 Rev. N/C: Limited to S/N 9145</td>
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<td>EASA.A.S.01585</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td>CAUK RESPONSIBILITIES</td>
<td>BOEING 737-200</td>
<td>Installation of Cargo Hold Smoke Detection &amp; Fire Suppression System</td>
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<td>EASA.A.S.01586</td>
<td>DASSAULT FALCON SERVICE</td>
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<td>FALCON 10</td>
<td>Update ATC Transponder for elementary surveillance DFS 1253/111A</td>
<td>See Change approval sheet DFS 1253 associated list of affected technical documents</td>
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<td>EASA.A.S.01589</td>
<td>DASSAULT FALCON SERVICE</td>
<td>FALCON 10</td>
<td>DFS1226 Revision D dated 9 June 2005: Installation of Collins TCAS-400w ith one or two TDR-940</td>
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<td>1. See Change approval sheet DFS 1226 associated list of affected technical documents2. AFM Supplement P10 DFS No 34 dist. 09.06.2005</td>
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<td>EASA.A.S.0159</td>
<td>TRANSAIRCO S.A.</td>
<td>PC-12, PC-12/45 and PC-12/47</td>
<td>Installation of a Garmin GMA-340 Audio Panel. Data required: (Engineering Order TSP/CD-23.50.0160).</td>
<td>- TSA/P/23.50.0160 Rev. A dated 10/AUG/05 and Rev. B dated 16/JAN/06 and all later approved revisions</td>
<td>1. KMA-24H Audio Panel is a pre-requisite installation. 2. This installation should not be incorporated in any aeroplanes unless it is determined that the interrelationship between this installation and any previously approved configuration changes will not introduce any adverse affect upon the aircraft.</td>
<td>03/07/2006</td>
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<td>EASA.A.S.01591</td>
<td>JET AVIONICS S.A.</td>
<td>CESSNA 650 Citation III</td>
<td>BDI 004-05 Updating to TCAS II &amp; ELT system and installation SSFDR</td>
<td>Master Drawing List No. BDI 004-05 Rev. 0 dated 18/AUG/2005</td>
<td>Applicable only if the STC No STB02CH of 16 may 1996 was previously installed</td>
<td>11/10/2005</td>
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<td>EASA.A.S.01594</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>MSN 498, 500, 502, 503, 516, MSN 423, 443, 448, 465, AIRBUS A330-203</td>
<td>Air France A330 “Ulysse” cabin reconfiguration in accordance with Change Approval Sheet No. CAAS-004-01 dated 16/09/2005 (or later EASA approved revision)</td>
<td></td>
<td>The approval of this STC is limited to Airbus A330-203 Serial Numbers 423, 443, 448, 458, 465, 481, 498, 500, 502, 503,516,519. The Limitations and conditions defined in the EASA Datasheet are applicable. A copy of this STC and the CAS must be enclosed in the files of each modified aircraft.</td>
<td>12/10/2005</td>
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<td>EASA.A.S.01596</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>BOEING 747-400</td>
<td>Air France B747-400 Big Bins modification. This is a revision to the EASA STC Ref 2004-4528, to extend the applicability to Boeing 747-400, F-REGOA, MSN 24996. The general description is included in the Air France Letter BL.NI 1093/EB and in the AEO-47-25-2071 Revision 2 Model: B747-400, MSN 24996...</td>
<td></td>
<td>1. The approval of this modification is limited to the Boeing B747-400 MSN 24996, 24990, 25344, 25660, 25661, 25662, 25663, 25668, 25669, 25671, 25251, 25520, 25559, 25569, 25599, 25569, 25599, 25608, 25615, 24154,24155. The limitations and conditions defined in the EASA Datasheet are applicable. 2. A copy of this STC and of the CAS must be enclosed in the files of each modified aircraft.</td>
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<td>EASA.A.S.01598</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>560</td>
<td>Installation of CAS67A (incl. MST67A), Mark VIII, Kannad 406AF and Garmin GPS400</td>
<td>Table of Contents FO E-05 (ASR 2005-042-00) Rev. 0 or later approved Revision/Fight Manual Supplement FMS- ASR2005-042-00 Rev. A (TCAAS CAS67A) or later approved Revision/Fight Manual Supplement FMS- ASR2005-042-01 Rev. A (XPDR MST67A) or later approved Revision/Fight Manual Supplement FMS- ASR2005-04...</td>
<td>Certification Basis of the Change: CS 25, additionally AMC 20-4 (Basic R-Nav), AMC 20-5 (GPS), JAA Leaflet No. 8, Rev. 2 (ACAS II), JAA Leaflet No. 12 (TAWS), JAA Leaflet No. 13 Rev. 1 (Mode-S ELS), JAA Leaflet No. 16 (Mode-S, EHS)</td>
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<td>GAMA ENGINEERING LTD.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BEECH 200</td>
<td>Major Avionic Upgrade</td>
<td>BEECH 200</td>
<td>Lees Avionics Ltd Modification HL/MOD/673</td>
<td>AFM Supplements AFMS673001, AFMS673002, AFMS673003 &amp; AFMS673004 at issue 1 dated January 23, 2004 or later approved revision.</td>
<td>09/04/2006</td>
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<td>EASA.A.S.01606</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>LOCKHEED L1011 - TRISTAR</td>
<td>Introduction of a stand-alone GPS receiver</td>
<td>PIPER PA-34-200</td>
<td>STC Twenty One AFM Supplement S21.FMS-0007 Issue 1 or later approved revision*</td>
<td>22/03/2006</td>
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<td>EASA.A.S.01612</td>
<td>SOGERMA SERVICES</td>
<td>REV 1</td>
<td>AIRBUS A340-642</td>
<td>This STC deals with the installation of 4 new double seats in Y class (Kyoto manufacturer). Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services.</td>
<td>AIRBUS A340-642</td>
<td>The associated Modification Approval Sheet is referenced SOG-25-1233 issue 1 dated 30/11/2005. All documents related to this STC are listed in the Master Data List Drawing n° MDL-25-1233, Issue No 8. (or later EASA approved revision). TECHNICAL REPERCUSSION SHEET reference 21-1233 issue B.</td>
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<td>EASA.A.S.01615</td>
<td>HAWKER BEECHCRAFT Ltd.</td>
<td>BAE 125-800 SERIES</td>
<td>Steep Approach Capability</td>
<td>Raytheon Aircraft Services Certification Plan ASIC-C-008-AV1253 Raytheon Aircraft Services AFM Supplements Flight ManualRef. HS 1.16 Supplement 10 Issue 1 &amp; HS1.15 Supplement 15 Issue 1 or later approved revisions.</td>
<td>25/10/2005</td>
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<td>CESSNA DUESSELDORF CITATIONBERVIC E CENTER GmbH</td>
<td>LEARJET 35A LEARJET 35</td>
<td>Installation of Flight Data Recorder F1000</td>
<td>Engineering Order 2003-4. Revision B or later EASA approved revisions.</td>
<td>26/10/2005</td>
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<td>EASA.A.S.01617</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td>BAE 146/AVRO 146 SERIES</td>
<td>Installation of BRNAV Compliant GPS</td>
<td>Bournemouth Aviation (Consultants) Modification BAC/146/064/issue1 or later approved revision Bournemouth Aviation (Consultants) AFM supplement FMS/353/31 Issue 1 dated 25 November 2005 or later approved revision.</td>
<td>12/12/2005</td>
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<td>EASA.A.S.01620</td>
<td>GAMA ENGINEERING LTD</td>
<td>BEECH 200, SUPER KING AIR</td>
<td>Major Avionic System Upgrade</td>
<td>AFMS577001 as x for GNX530B BRNav complianceAFMS577002 as x for GNX6430 BRNav complianceAFMS577003 as x for GNX3300 Mode S</td>
<td>19/12/2005</td>
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<td>AIM AVIATION (HENSHALLS) LTD</td>
<td>BOEING 737-400</td>
<td>Installation of Galley 1 &amp; Galley 2 for Future International Airways</td>
<td>DD 1037: issue new or later approved issue</td>
<td>06/01/2006</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>BOEING 737-300 SERIES</td>
<td>Installation of new Recaro Seating convertible and standard seats and cabin refurbishment to PG-design on Lufthansa’s Boeing B737-300 aircraft</td>
<td>&quot;Certification Program CP25-05/23M, issue 01, dated 01.02.2006 Supplemental Type Definition Document STDD025-05/23M Iss.01, Rev. 00, 27.02.06&quot;</td>
<td>Certification Basis of the change: FAR25 as defined in FAA TCDS A116WE. The applicant showed compliance to CS 25.076(9)(2), CS 25.811 (d), (e), CS 25.812, CS 25.085(3)(a), (b), (c),(f).</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>BEECH A36</td>
<td>Replacement of existing KI-525A HSI system with new KI-825 EHSI system</td>
<td>AFM Supplement Doc. 2629189-AM-01, issue 2, December 2005, or later EASA approved issue</td>
<td>Certification Basis of the change: FAR25 as defined in FAA TCDS A116WE. The applicant showed compliance to CS 25.076(9)(2), CS 25.811 (d), (e), CS 25.812, CS 25.085(3)(a), (b), (c),(f).</td>
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<td>EASA.A.S.01627</td>
<td>AIR FRANCE</td>
<td>BOEING 767-300</td>
<td>Cabin Reconfiguration with an increased seating configuration for LOT Polish Airlines.</td>
<td>Air France Change Approval Sheet CAS-B767-06-006 Revision 0 or later approved revision Air France AMF Supplement MA.PM 133/S6 KA Initial Release dated 2006-05-17 or later approved revision.</td>
<td>24/04/2006</td>
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<td>EASA.A.S.01631</td>
<td>ATIS INTERNATIONAL</td>
<td>DC-10 SERIES</td>
<td>Installation of a Honeywell Enhanced Ground Proximity Warning System MK VII.</td>
<td>&quot;Certification Program CP25-05/23M, issue 01, dated 01.02.2006 Supplemental Type Definition Document STDD025-05/23M Iss.01, Rev. 00, 27.02.06&quot;</td>
<td>Certification Basis of the change: FAR25 as defined in FAA TCDS A116WE. The applicant showed compliance to CS 25.076(9)(2), CS 25.811 (d), (e), CS 25.812, CS 25.085(3)(a), (b), (c),(f).</td>
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<td>EASA.A.S.01635</td>
<td>SOGERMA SERVICES</td>
<td>A340-642</td>
<td>This STC deals with the installation of a 2 new triple seats in W class (Contour design seats) Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services.</td>
<td>&quot;Certification Program CP25-05/23M, issue 01, dated 01.02.2006 Supplemental Type Definition Document STDD025-05/23M Iss.01, Rev. 00, 27.02.06&quot;</td>
<td>Certification Basis of the change: FAR25 as defined in FAA TCDS A116WE. The applicant showed compliance to CS 25.076(9)(2), CS 25.811 (d), (e), CS 25.812, CS 25.085(3)(a), (b), (c),(f).</td>
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<td>Limitations/Conditions</td>
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<td>AERODISA S.L.</td>
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<td>EMBRAER EMB-120</td>
<td>Structural Permanent element for a Winch installation.</td>
<td>1. Doc. AED2005-02M_MDL &quot;Master Data List&quot;, Ed. 3 Rev. 0, dated 5 October 2008; 2. Doc. AED2005-02M_D02 &quot;Airworthiness Manual Supplement&quot;, Ed. 3 Rev. 0, dated 5 October 2008.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third party....</td>
<td>04/11/2008</td>
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<td>EASA.A.S.01641</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
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<td>CONSTRUCCIONES C-212</td>
<td>Installation of CAS 87A ACAS II and dual MST 87A Mode-S XPDR</td>
<td>Engineering Order, Report 2005-37, Rev. B, dated 17.01.2006; Airplane Flight Manual Supplement, Report 2005-37-8, Rev. A, dated 23. December 2005</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>23/02/2006</td>
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<td>EASA.A.S.01651</td>
<td>BRITISH AIRWAYS PLC t/a BRITISH AIRWAYS ENGINEERING</td>
<td></td>
<td>BOEING 737-400</td>
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<td>Installation of Enhanced Mode S</td>
<td>British Airways Engineering Project Certification Plan Ref.CP-0002-04 Issue 1 or later approved revision</td>
<td>None</td>
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<td>25/11/2005</td>
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<td>EASA.A.S.01653</td>
<td>PHOENIX AEROSPACE LIMITED</td>
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<td>JETSTREAM 3202</td>
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<td>Installation of Dual GNS530 with Dual GI-106A &amp; Dual KNI-582 RMI</td>
<td>Phoenix Aviation Design Services Master Document List PACS/ML/100300/01 Phoenix Aviation Design Services AFM Supplement PACS/FMS/JS1-JS2/0001 Revision 0 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>10/02/2006</td>
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<td>EASA.A.S.01658</td>
<td>AIR FRANCE</td>
<td></td>
<td>BOEING 767</td>
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<td>Emergency Escape Path Marking System</td>
<td>Air France Industries Modification EA33-8767-05-00B Revision 00 or later approved revision Air France Industries AFM Supplement reference MA.PM 827/05 CE Initial Release dated 06.06.2005 or later approved revision.</td>
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<td>EASA.A.S.01665</td>
<td>TATENHILL AVIATION Ltd.</td>
<td>CAAS UK RESPONSIBILITY</td>
<td>PILATUS PC-12/45</td>
<td>Substantiation of Garmin AT GPS400 installation for IFR</td>
<td>Tatenhill Aviation Ltd Modification TAL_TAD 019/05 TatenhillAviation AFM Supplement TAL-BRNAV/AFS-TAD 019/05 Issue 1 or later approved revision.</td>
<td>Certification Basis of Change: FAR 25 as defined in TCDS A10CE issue 05, additionally JAA Leaflet No. 13, Rev. 2 (ACAS II), JAA Leaflet No. 13, Rev. 1 (Modes S ELS)</td>
<td>Issue 1 or later approved revision.</td>
<td>20/06/2006</td>
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<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>25C</td>
<td>Collins TCAS-4000 with Dual TDR-94D System Installation</td>
<td>Engineering Order No.: 212-LR Rev. IR with Appendix Airplane Flight Manual Supplement AG524-071, Revision IR</td>
<td>Certification Basis of Change: FAR 25 as defined in TCDS A10CE issue 05, additionally JAA Leaflet No. 13, Rev. 2 (ACAS II), JAA Leaflet No. 13, Rev. 1 (Modes S ELS)</td>
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<td>DASSAULT FALCON SERVICE</td>
<td>MYSTERE FALCON 2000EX</td>
<td>MYSTERE FALCON 2000</td>
<td>DFS D078 dtd. 5.4.2006: EGPWS Software 222-222 see Change approval sheet DFS D078 and associated list of affected technical documents</td>
<td>1. Activation of Runway Awareness Advisory System (RAAS) is not approved through this STC. 2. This STC is approved only for the configuration as defined in Mod Sheet DFS D078. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>CRANFIELD AEROSPACE LTD</td>
<td>CAA UK RESPONSIBILITY</td>
<td>JETSTREAM 4100</td>
<td>Installation of a Honeywell Global GNS-XLS</td>
<td>AFM Supplement - Cranfield Supplement No 2 Issue 1 (Type Design Definition Addendum CA/DDE2184 Issue 4DF) Jetstream41/1232</td>
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<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>HS.125 SERIES 700B</td>
<td>Installation of Honeywell EGPRS Mark VIII System and interface to Honeywell KMD-850 MFD via a KAC-502 EGPRS Module</td>
<td>Installation in accordance with Aero-Dienst Engineering Order ref 205-HW, Rev. IR, dated 06. Feb. 2006, or later approved revision. To be operated in accordance with Aero-Dienst Airplane Flight Manual Supplement ref AG516-071, Rev IR, dated10 April 2006, or later approved revision.</td>
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<td>BBJ DESIGN SERVICES LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>767-200</td>
<td>Replacement of All G4B Galley.</td>
<td>BBJ Design Services Modification BBJ302</td>
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<td>27/06/2006</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>M20E</td>
<td>BRNAV capacity with AP Steering</td>
<td>Modification No. 0205/ AerotecnAviation Flight Manual Supplement BRNAV/Garmin GNS 430 - edition 1 dated 12 October 2005</td>
<td>Modification limited to s/n 28R7031294 To use AFMS BRNAV/Garmin 430 must be used with this modification</td>
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<td>EASA.A.S.0168</td>
<td>SOGERMA SERVICES</td>
<td>REV. 2</td>
<td>A330-304, MSN 537 AND 586</td>
<td>Traffic Alert and Collision Avoidance System, TCAS change 7 software and two ATC Mode S installations in accordance with Modification Approval Sheet SOG 34-87 Issue 2 dated 08 July 2004</td>
<td>1</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configuration shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked...</td>
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<td>TENENCIA Ltd.</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>BAE JETSTREAM 3100</td>
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<td>DD/AA/397 AAE Supplement 4 issue 1 or later approved revision to Approved AFM Ref. HP4.11</td>
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<td>AEROSPACE DESIGN AND ENGINEERINGCONSULTANTS Ltd</td>
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<td>737</td>
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<td>Aerospace Design &amp; Engineering Consultants Ltd Modification Summary MOD-0023</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV. 1</td>
<td>BOEING 767</td>
<td>Cabin Reconfiguration from Eva Air to Aeroflot Standard</td>
<td>1</td>
<td>FAA STC SA5950NM (Door 2 Rear Galley Installation) is not included in this approval.</td>
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<td>STEINBEIS GmbH &amp; Co. KG FÜER TECHNOLOGIETRANSFER</td>
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<td>DG LS8-T</td>
<td>LS8-In Hydraulic extension system/Exchange of the electric driven engine extension system to a hand-driven hydraulic system</td>
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<td>Change of pages in the Flight and Maintenance Handbook. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>none</td>
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<td>EASA.A.S.01696</td>
<td>JET AVIATION GmbH</td>
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<td>LEARJET 31</td>
<td>Installation of Bluesky C-1000 SATCOM</td>
<td>1</td>
<td>Jet Aviation Modification 2004-7 Revision A dated November 14, 2005 or after approved revision</td>
<td>none</td>
<td>03/08/2006</td>
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<td>BOEING 767-300 SERIES BOEING 767-200 SERIES</td>
<td>TCAS II Change 7 System Installation</td>
<td>1</td>
<td>ATS International Master Data List 0267-01-A-2103-R00 or later approved revision</td>
<td>ATS International AFM Supplement 0267-01-A-2301-R03 or later approved revision</td>
<td>22/05/2006</td>
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<td>Associated Technical Documentation</td>
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<td>Date of Issue</td>
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<td>EASA.A.S.01699</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>BEA9NG 737-500</td>
<td>Installation of new Recaro Aircraft Seating convertible and standard seats and cabin refurbishment to PG-design on Lufthansa's Boeing 737-500 aircraft.</td>
<td>Certification Program CP25-06/20M, issue 01, dated 09.03.2006Supplement Type Definition Document STD25-05/20M Iss. 1, 18.04.2006Compliance Record List CRL25-05/25M Iss. 1, 15.04.2006Applicant's Declaration of Compliance ADc dated 20.04.2006</td>
<td>Certification Basis of the change: FAR25 as defined in FAA TCDS A16W.</td>
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<td>EASA.A.S.01701</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>BEECH A36</td>
<td>Title: GARMIN GNS 430/530 upgrade for GPS approachDescription: Upgrade of existing GNS 430/530 for GPS approach and installation of annunciator panel</td>
<td>Aircraft Flight Manual Supplement. Doc. 2031235-AM-01, issue 1 or later EASA approved issue</td>
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<td>EASA.A.S.01702</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>CESSNA 550 CITATION II</td>
<td>Honeywell EGW5 Mk VIII</td>
<td>Change Notice No: 170B-CN Rev. N/C or later approved Revision</td>
<td>Flight Manual Supplement TAL-IRNAV FMS-TAO 062/06 Issue 1</td>
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<td>SABENA TECHNICS BGC</td>
<td>DC-9-83 (MD-83) DC-9-82 (MD-82)</td>
<td>Elementary and Enhanced Surveillance Installation</td>
<td>MLD 0269-01-A2103-R06AFMS 0269-01-A2301-R05</td>
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<td>TATENHILL AVIATION Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>CESSNA F182-Q</td>
<td>Design data and equipment installation in order to approve the aircraft for IFR.</td>
<td>Flight Manual Supplement TAL-IRNAV FMS-TAO 062/06 Issue 1</td>
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<td>08/01/2007</td>
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<td>FLYBE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
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<td>Introduction of Class E Compartement Cargo Layout</td>
<td>FlyBe Modification Statement BE/MOD/002/06 Issue 1</td>
<td>Restricted to Aircraft Serial Number 2042 &amp; 2052</td>
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<td>EASA.A.S.01713</td>
<td>TURNTIME TECHNOLOGIES AB</td>
<td>AIRBUS A321 AIRBUS A320 AIRBUS A319</td>
<td>Installation of Sliding Carpet Loading System in Airbus A319/320/321 in accordance with Telair International AB Master Document List (MDL)</td>
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<td>FR AVIATION LTD D/B/A COBHAM AVIATION SERVICES UK</td>
<td>CAA UK RESPONSIBILITY</td>
<td>FALCON 20 BASIC, D AND E SERIE</td>
<td>DC-0052 std. 7.6.06: Introduction of TAWS Display</td>
<td>Design Change Statement DC-0052 std. 7.6.05-7.6.06 AFM Supplement FRA/FM809 std. 19.5.06-19.5.07, associated list of affected technical documents</td>
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<td>EASA.A.S.0173</td>
<td>SOGERMA SERVICES</td>
<td>REV. 1</td>
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<td>AIRBUS A340-642</td>
<td>Installation in upper class side facing Britia Mark 2 seat equipped with the inflatable restraints, monuments, a bar and a chandelier according to MAS SOG-25-597 issue 4 dated 02/07/04 and reference document: Aircraft Flight Manual A340-600 1.02.09 P.01. Supplementary Structural Repair Manual...</td>
<td>STC Twenty One Modification S21.34.39-1014AFM</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft engine configurations shall be determined by the installer.</td>
<td>01/01/2008</td>
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<td>EASA.A.S.01739</td>
<td>STC TWENTY ONE LIMITED</td>
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<td>SHORTS SD3-60</td>
<td>Introduction of ACAS II &amp; Single Mode S Transponders with Elementary Surveillance.</td>
<td>STC Holder CAA UK RESPONSIBILITY</td>
<td>1. The installation is compatible with the installation of a VHF antenna between Fr 51 and Fr 52, and a Canadian Marconi Aero-H+ antenna between Fr 58 and Fr 61 (separately or simultaneously). The installation requires both ACT 1 and 2 be installed on A320 (pre-mod) identical to MSN 2165).</td>
<td>13/03/2006</td>
<td>Active</td>
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<td>EASA.A.S.01740</td>
<td>AIRBUS S.A.S.</td>
<td>REV. 1</td>
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<td>A320 MODELS</td>
<td>SAT TV Tailwind 550 antenna installation in accordance with MAS SOG-53-1252 issue 2 dated 27/07/06</td>
<td>MPD Supplement MOD SOG-53-1252 dated Feb 01/06</td>
<td>1. The installation is compatible with the installation of a VH3 antenna between Fr 51 and Fr 52, and a Canadian Marconi Aero-H+ antenna between Fr 58 and Fr 61 (separately or simultaneously). 2. The installation requires both ACT 1 and 2 be installed on A320 (pre-mod) identical to MSN 2165).</td>
<td>30/06/2008</td>
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<td>FOCA STC 23-15-001 ISS2 PILOTUS PC-12/47 PILOTUS PC-1245 PILOTUS PC-12</td>
<td>Aircell 3100 Iridium SATCOM Installation</td>
<td>A&amp;M Avionics PC No. 02/2006 Rev. 0, dated 30/11/2005.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft engine configurations shall be determined by the installer.</td>
<td>07/06/2006</td>
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<td>EASA.A.S.02470</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING (factory for Cessna 208 models)</td>
<td>208, 208B (208B NOT VALIDATED)</td>
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<td>Mount assembly for external side loads</td>
<td>AFM supplement AFSS-002700101_MBO_03 22006Instructions for continued airworthiness ICA_D02700101_MBO_03 22006</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft engine configurations shall be determined by the installer.</td>
<td>07/06/2006</td>
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<td>EASA.A.S.02474</td>
<td>SOGERMA SERVICES</td>
<td>REV 1</td>
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<td>DC-9-21</td>
<td>Modification Janus Phase I: Cancellation of the restricted flights in icing conditions note. This revision reflects the transfer of this STC from EADS Soergema Services to Soergema Services.</td>
<td>Modification Approval Sheet (MAS), ref. SOG-520-000, issue 3 and dated April 12, 2005Aircraft Flight Manual Supplement, ref. BD-1251/01, issue 17, dated April 11, 2005.</td>
<td>1. This STC is limited to the MSN 47308.</td>
<td>04/01/2008</td>
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<td>RUANG AEROSPACE SERVICES GmbH</td>
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<td>CESSNA 550 CITATION II</td>
<td>ACAS II 5CASATDA with Dual Mode S XPR/MST 67A-EHS</td>
<td>Change Notice No. TDSA-CN Rev. NC or later approved Revision: Airplane Flight Manual Supplement 170-AAM/PBS Rev. NC or later approved Revision</td>
<td>Certification Basis of the Change: FAR 25 as defined in TCDS A22DC: Rev. 58; additionally JAA Leaflet No. 8, Rev. 2 (ACAS II), JAA Leaflet No. 13, Rev. 1 (Mode-S ELS) and JAA NPA 20-12a (Mode-S EHS)</td>
<td>19/04/2006</td>
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<td>FR AVIATION LTD, B \ BBA</td>
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<td>DORNIER 288-202K</td>
<td>Operational Upgrade</td>
<td>FR Aviation Design Change: DC-005: Issue 1 or later approved revision: FR Aviation AFM Supplement: FRAHFM/04 Revision 0 or later approved revision. This STC is approved only for the product configuration as defined in the approved design data referenced in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or amended.</td>
<td>24/01/2006</td>
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<td>PHOENIX AEROSPACE LIMITED</td>
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<td>BEECH A36</td>
<td>Installation of Skywatch Traffic Avoidance</td>
<td>Phoenix Aviation Design Services Modification: PAADS_10141Phoenix Aviation Design Services OMF Supplement: PAOMS/35/001 Revision 0 dated March 2006 or later approved revision</td>
<td>07/06/2006</td>
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<td>PIPER PA-31T</td>
<td>Modification SIBA MD01_2006: Replacement of one ATC mode C by Mode S and replacement of an old DME with adding an ATC antenna on the top of frame</td>
<td>None</td>
<td>This STC is limited to: S/N 317-793/004 F-GPBF</td>
<td>31/08/2006</td>
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<td>CESSNA 550 (CITATION V)</td>
<td>CESSNA 550 (CITATION V)</td>
<td>Sandel ST3400 TAWS installation left hand and right hand</td>
<td>Engineering Order 2631660-EO-01 Issue 1 or later approved Revisions: Aircraft Flight Manual Supplement 2631650-AM-01 Issue 4 or later approved Revisions: Aircraft Flight Manual Supplement 2631650-AM-01 B Issue 1 or later approved Revisions</td>
<td>02/04/2006</td>
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<td>DFS 2568 distr. 21 Dec 05; DFS AFM Supplement F20 51 to DTM 589/ DTM58/DTM569</td>
<td>see Change approval sheet DFS 25686: DFS AFM Supplement F20 51 to DTM 589/DTM58/DTM569 anti- associated list of affected technical documents</td>
<td>Certification Basis of the Change: CS 25, additionally JAA Leaflet No. 12 (TAWS)</td>
<td>03/07/2024</td>
<td>Page 1.510 of 1,892</td>
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<td>BOURNEMOUTH AVIGATION CONSULTANTS LTD.</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>BAE 146-200</td>
<td>Layout of Passenger Accommodation (LOPA) for B4, B5, B6 &amp; B7</td>
<td>Bournemouth Aviation (Consultants) Ltd Modification BAO146096</td>
<td>19/04/2006</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING 737-400</td>
<td>Installation of 136 (16/120) PAX LOPA</td>
<td>Bouremouth Aviation (Consultants) Ltd</td>
<td>Modification BAC/373/049</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engineconfigurations shall be determined by the installer.</td>
<td>07/04/2006</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BAE ATP</td>
<td>Introduction of Rockwell Collins ACAS II and Dual Mode S Transponder System</td>
<td>STC21 Change S21.34-23-1008STC21 AFM Supplement S21.FMS-ATP-0002 issue 1 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engineconfigurations shall be determined by the installer.</td>
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<td>SLINGSBY ADVANCED COMPOSITES LTD.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BEECH 76</td>
<td>Introduction of Bendix/King KCS 55A Compass, KX165 NAV/COM,KR87 ADF, King KN62A DME &amp; KI229 Systems</td>
<td>Singsbys Aviation Ltd Modification SAL/MOD/BEE76/01Singsbys Aviation Ltd AFM Supplement Singsbys Beach 76-01 Issue 1 or later approved revision Singsbys Aviation Ltd AFM Supplement, Singsbys Beach 76-02 Issue 1 or later approved revision.</td>
<td>20/06/2006</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>CESSIONA 550 CITATION II</td>
<td>Introduction of an AirCell Satellite Telephone System ST 3100 to the aircraft.This modification installs the handset into an arm rest on the near starboard side of the aircraft. The Satellite Phone Transceivers installed in the mid cabin drink stowage on the port side of the aircraft. A circuit br...</td>
<td>Installation Accomplishment Instruction, Serial No : S21.ACC-0026, Iss. 1Aircraft Maintenance Manual, Wiring Diagram Manual, Illustrated Parts Catalogue, refer to STC Twenty One Ltd Supplemental Manual S21.TEC-0032 Drawing No. Title18-29-10004 Satcom Places3832-29-10002 Satcom Handle.</td>
<td>This design change should be embodied concurrently with or after the embodiment of STC Twenty One Limited minor design change S21.23-29-1000.</td>
<td>09/06/2006</td>
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<td>EASA.A.S.02500</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING 767-200-300</td>
<td>Introduction of SSR Mode S Enhanced Surveillance System</td>
<td>Master Document List S21.MDL-0049 AFM supplement S21.FMS-0028 Issue 1</td>
<td>Boeing 767-200/300 series. It is a condition of this STC that transponders of different manufacturers must not be intermixed, refer to STC 21 IPC supplement S21.TEC-0033 for further details.</td>
<td>12/10/2007</td>
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<td>EASA.A.S.02501</td>
<td>AVIONIK STRAUSSING ENTWICKLINGS GmbH</td>
<td>PA-46-310P</td>
<td>Installation of GNS530, with GDC31 Roll Flattening Converter</td>
<td>FMS ASR 2006-004-01 Rev. 1, Garmin GNS500/600/605 ASR 2006-004-02 Rev. 1, DAC GDC31</td>
<td>05/04/2006</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>A320-200</td>
<td>A320 Pax Capacity Increase from 156 to 168 Pax</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>BOEING 737-900</td>
<td>KLM 737-800, -900 Recaro Seats Installation</td>
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<td>- Certification Programme CP-B737-05-001 Rev2, March 28th 2006. - Compatibility of this design change with other previously approved modifications must be determined by the installer. - The installation is limited to the B 737-800, -900 listed in the CP-B737-06-001 Rev2, March 28th 2006.</td>
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<td>EASA.A.S.02505</td>
<td>THEILERT AIRCRAFT ENGINES GmbH</td>
<td>CESSNA 172</td>
<td>Installation of Universal Fuel System</td>
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<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configuration shall be determined by the installer.</td>
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<td>07/02/2006</td>
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<td>AIM AVIATION (HENSHALLS) LTD</td>
<td>BOEING 747-428</td>
<td>Installation of Upper Deck Galley partnr 747TASAO/A10/0406 into a B747-428 passenger to cargo conversion (747-TASAO) as defined in STC Definition Document DD 1022 dated 01-05-2006 or later EASA approved revisions.</td>
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<td>- AIM Henshalls Component Maintenance Manual 25-36-76- AIM Henshalls Illustrated Parts List 25-36-76</td>
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<td>03/05/2007</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>BOEING 767-300</td>
<td>cabin reconfiguration</td>
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<td>Air France Change Approval Sheet No EA33- B767-06-001 rev 0</td>
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<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>BOMBERDIER BD-3- SHERPA BOMBERDIER BD-3-60- SHERPA</td>
<td>Collins TCAS 4000 with dual TDR-94D Transponder Installation</td>
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<td>Engineering Order 2036479-EO-01 Issue 2, dated 19.05.2006 or later approved Revisions</td>
<td>Aircraft Flight Manual Supplement 2036479-AM-01A Issue 1, dated January 2006 or later approved Revisions</td>
<td>01/08/2006</td>
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<td>EASA.A.S.02517</td>
<td>AEROSPAC DESIGNS ENGINEERING CONSULTANTS LTD</td>
<td></td>
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<td>737-4YD</td>
<td>Introduction of 6/150 Passenger Configuration and Associated Customisation</td>
<td>- Modification Summary: MOD-0026 Issue C dated 7th September 2006, or later approved revision - Modification Certification Plan: MDP-0026 Issue C dated 7th September 2006, or later approved revision.</td>
<td>Compatibility of this design change with other previously approved modifications must be determined by the installer.</td>
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<td>20/10/2006</td>
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<td>EASA.A.S.02521</td>
<td>THOMSONFLY LTD (FORMERLY BRITANNIA AIRWAYS LTD.)</td>
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<td>BOEING 767-300</td>
<td>Introduction of STG Aerospace Photo-Luminescent Passenger Emergency Escape Path Lighting System</td>
<td>- Thomsonfly Ltd MOD/767/1463 Issue 2 or later approved revision STG Aerospace AFM Supplement M40-11 Revision A dated August 20 2002 or later approved revision</td>
<td>Cabin &amp; Exit Lighting levels must be measured for acceptability on each installation in accordance with Thomsonfly Technical Engineering Order TEO/767/1463 Issue 2 dated 26 April 2006 or later approved revision.</td>
<td></td>
<td>02/05/2006</td>
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<td>EASA.A.S.02522</td>
<td>ATS INTERNATIONAL</td>
<td>REV 1</td>
<td>A318/A319/A320/A321 ALL MODELS</td>
<td></td>
<td>Mode 5 Elementary and Enhanced surveillance</td>
<td>Master Data List (MDL) ref. 0264-01-A-2103 revision 06, dated June 19, 2006 of later EASA approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.A.S.02524</td>
<td>SOCIETE AIR FRANCE S.A.</td>
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<td>BOEING 777-200</td>
<td>Air France B777-200 &quot;Olyssia&quot; cabin modification with Contour (CAS B777-04-003) or BEA business class seats (CAS B777-05-015).</td>
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<td>03/11/2006</td>
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<td>EASA.A.S.02525</td>
<td>BINDER MOTORENBAU GmbH</td>
<td></td>
<td>ASH 25 E</td>
<td></td>
<td>Modification of outer wing panel with winglets and 27 m wingspan</td>
<td>ASH 25 E Technische Mitteilung Nr. 18/01</td>
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<td>EASA.A.S.02526</td>
<td>SOCIETE AIR FRANCE S.A.</td>
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<td>AIRBUS A320-211</td>
<td></td>
<td>Photo-luminescent EEPm system installation</td>
<td>Change Approval Sheet (CAS) ref. A320-06-004 revision, dated May 24, 2006 or later EASA approved revision</td>
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<td>EASA.A.S.02527</td>
<td>HAWKER BEECHCRAFT LTD</td>
<td></td>
<td>BAE 125 SERIES 800B</td>
<td></td>
<td>To introduce completion of the upgrade to the existing Terrain Awareness Warning System (TAWS) installation to include terrain display.</td>
<td>Raytheon Aircraft Service Modification SOC/125Raytheon (AFMSup p H. S. 115 Amendment No. 18)</td>
<td>Applicable to Aircraft Serial number 258208 only</td>
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<td>02/05/2006</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>EASA.A.S.02530</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td>CESSNA S550 (CITATION S/II) CESSNA S500 (CITATION V) CESSNA S500 CITATION II CESSNA 500 CITATION</td>
<td>Installation of a Garmin GNS 530A and a Honeywell Marker Beacon KR21, for EU basic RNAV operations</td>
<td>System Description 2032044-DE Iss. 2, Drawing List 2032044-DL Iss. 2, MoC list 2032044.AFM Supplements 2032044-Am-6 Iss. 3 and 203044AM-B Iss. 2</td>
<td>None</td>
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<td>13/06/2006</td>
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<td>10174420</td>
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<td>EASA.A.S.02531</td>
<td>KUERZI AVIATION AG</td>
<td></td>
<td>LEARJET 60</td>
<td>Installation of SSFDR and MFDAU Installation of a Honeywell Solid State Flight Data Recorder and Teledyne Mini Data Acquisition Unit in Learjet Model 60</td>
<td>Master Document List No. MDL-200514764</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer.</td>
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<td>11/12/2006</td>
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<td>EASA.A.S.02532</td>
<td>BINDER MOTORENBAU GmbH</td>
<td></td>
<td>ASW 22 BLE</td>
<td>Binder Motorenbau Technische Mitteilung ASW 22 BLE B2Changes in engine installation:- Different type of spindle drive- Installation of a primer ball- CFRP gear mount</td>
<td>Binder Motorenbau Technische Mitteilung ASW 22 BLE B2drawings and Manual pages according to Tech. Note.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer.</td>
<td>-</td>
<td>12/04/2006</td>
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<td>EASA.A.S.02533</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV 1</td>
<td>AIRBUS A340-300</td>
<td>Installation of CAIR System in A340-300</td>
<td>-</td>
<td>-</td>
<td>-</td>
<td>18/08/2006</td>
<td>Active</td>
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<td>EASA.A.S.02534</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td></td>
<td>BOEING 757-200</td>
<td>Layout of Passenger Accommodation - 92 Business &amp; 8 Economy/Crew</td>
<td>Bournmouth Aviation (Consultants) BAC/757/006</td>
<td>-</td>
<td>-</td>
<td>20/03/2006</td>
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<td>EASA.A.S.02536</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>F27 MARK 500</td>
<td>Change of WXR to RDR 2000 Radar</td>
<td>Data package containing the following documents:1. Fokker Services SB F27/28-70: Fuel - Introduction of Fuel Alcs andCDCCLs for the F27M200300440M0000 0700.2 Fokker Services Report SE-671. Fokker 50/60 Fuel Alcs and CDCCLs These reports are based on the fuel tank safety review performed by Fokker.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked.</td>
<td>-</td>
<td>04/08/2006</td>
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<td>EASA.A.S.02537</td>
<td>AERODISA S.L.</td>
<td>MYSTÈRE-FALCON 20GF</td>
<td>AED2006-01M TCAS II Installation for Falcon 20.</td>
<td>-</td>
<td>-</td>
<td>-</td>
<td>18/07/2007</td>
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<td>EASA.A.S.02538</td>
<td>Sogema Services</td>
<td>Airbus A320-211</td>
<td>Cabin modification from 150 to 167 Y/Chire. This revision reflects the transfer of this STC from EADS Sogema Services to Sogema Services.</td>
<td>Modification of approval sheet (MABS ref. DSO-25-1250 issue 1, dated April 14, 2006 or later EASA approved revision.</td>
<td>STC Twenty One Modification S21.34-25-1017/STC Twenty One AFM Supplement S21.25-0116 issue 1 or later approved revision.</td>
<td>REV 1</td>
<td>04/12/2008</td>
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<td>EASA.A.S.02539</td>
<td>STC Twenty One Limited</td>
<td>CAE UK</td>
<td>Installation of Two L Band Antennae</td>
<td>Active</td>
<td>Installation of Two L Band Antennae</td>
<td>REV 1</td>
<td>22/05/2006</td>
<td>Active</td>
<td>10014477</td>
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<td>EASA.A.S.02540</td>
<td>Bournemouth Aviation (Consultants) Ltd.</td>
<td>CESSNA 441 Series</td>
<td>Installation of Two L Band Antennae</td>
<td>Installation of Aircraft Registered within EASA Member State only</td>
<td>Bournemouth Aviation (Consultants) Definition Document ref J37/00/01 issue 1 dated March 08 2006</td>
<td>REV 1</td>
<td>17/05/2006</td>
<td>Active</td>
<td>10014475</td>
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<td>EASA.A.S.02542</td>
<td>AIM Aviation (Henshalls) Ltd.</td>
<td>BOEING 767-300</td>
<td>Installation of Galley F4 Part Number 7672FCF4040</td>
<td>Restricted to Aircraft Serial Number B76730MP-003 &amp; B76740</td>
<td>AIM Aviation (Henshalls) Change Number, HEN 1035</td>
<td>REV 1</td>
<td>29/06/2006</td>
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<td>EASA.A.S.02545</td>
<td>Kuerzi Aviation AG</td>
<td>Cessna 340A</td>
<td>Avionic Retrofit</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engineconfigurations shall be determined by the installer.</td>
<td>Master Document/Drawings List, Doc No. MDL-200514568 Rev B dated 18.04.2006AFMS, Doc No. FMS-200514568, Original dated 06.03.2006</td>
<td>REV 1</td>
<td>25/04/2006</td>
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<td>EASA.A.S.02547</td>
<td>CTT Systems AB</td>
<td>BOEING 767-300</td>
<td>Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via CAE NL STC SA009NL due to the potential for contamination of the dryer unit with dry contaminants in the dryer unit. A new dryer unit has been developed which removes this issue...</td>
<td>SB 300-000-01-117/AD Initial / SB 300-000-01-27-01/AD Initial/EASA AD 2006-0140-E or later approved revisions/Note: This revision reflects an amendment to the installation instructions and supersedes STC EASA.A.S.02547/Revision 1.</td>
<td>Installation of Aircraft Registered within EASA Member State only</td>
<td>REV 2</td>
<td>14/04/2009</td>
<td>Active</td>
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<td>EASA.A.S.02549</td>
<td>Premair Aircraft Engineering</td>
<td>BEECH 390</td>
<td>Satellite Telephone Installation</td>
<td>Premier Aircraft Engineering Modification 06-001</td>
<td>Premier Aircraft Engineering Modification 06-001</td>
<td>REV 1</td>
<td>11/05/2006</td>
<td>Active</td>
<td>10014487</td>
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<td>EASA.A.S.02552</td>
<td>Binder Motorenbau GmbH</td>
<td>ASH 25</td>
<td>Modification of outer wing panel with a 25 m wingspan</td>
<td>ASH 25 Technische Mitteilung Nr. 17B1</td>
<td>ASH 25 Technische Mitteilung Nr. 17B1</td>
<td>REV 1</td>
<td>03/02/2006</td>
<td>Active</td>
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<td>EASA.A.S.02555</td>
<td>MT-Propeller Entwicklungs GmbH</td>
<td>DIAMOND H-36 DIMONA</td>
<td>Installation of 2-blade MTV-1-A/1 180-03 constant speed propeller</td>
<td>For all serial numbers.</td>
<td>Umwelt- und Wartungsanweisung E-1433, including Anhang zum Flughandbuch E-1432</td>
<td>REV 1</td>
<td>12/12/2006</td>
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<td>EASA.A.S.02560</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>DORNIER DO 228-212, DORNIER DO 228-201, DORNIER DO 228-200, DORNIER DO 228-101, DORNIER DO 228-100</td>
<td>Installation of Sandel ST3400 TAWS Installation of Installation/Installation of Class A TAWS/RMI Indicator in the right hand instrument panel with a remote Annunciator and Switching unit on the left hand instrument panel. Single 3&quot; indicator is used to show compliance</td>
<td>AFMS 0092-001AFMS 0092-002AFMS 0092-003AFMS 0092-004</td>
<td>AFMS 0092-001 or later approved revision</td>
<td>22/06/2006</td>
<td>Active</td>
<td>10014198</td>
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<td>EASA.A.S.02561</td>
<td>TENENCIA Ltd.</td>
<td>A320</td>
<td>Cabin Reconfiguration in bi-classes on LAN A320</td>
<td>Tenencia Modification TEN 460ESA approved Tenencia AFM Supplement 2 Issue 1 or later approved revision</td>
<td>Tenencia Modification TEN 460ESA approved Tenencia AFM Supplement 2 Issue 1 or later approved revision</td>
<td>23/10/2006</td>
<td>Active</td>
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<td>EASA.A.S.02569</td>
<td>EADS SOGERMA</td>
<td>DC-9-21</td>
<td>JANUS Phase II modification</td>
<td>Eddy current inspections ref. MM085-337210 and MM805-337110A as defined in the Maintenance Schedule Supplement ref. MO 01-152 issue A, dated February 10, 2006.This STC is limited to the DC-9 MSN 47308.This STC is approved only for the product configuration as defined in the approved design datadef...</td>
<td>Eddy current inspections ref. MM085-337210 and MM805-337110A as defined in the Maintenance Schedule Supplement ref. MO 01-152 issue A, dated February 10, 2006.This STC is limited to the DC-9 MSN 47308.This STC is approved only for the product configuration as defined in the approved design datadef...</td>
<td>14/12/2006</td>
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<td>EASA.A.S.02570</td>
<td>GAMMA ENGINEERING Ltd.</td>
<td>BEECH 200</td>
<td>Mode S Transponder Installation</td>
<td>Form 1 Modification Statement LN/AM/706 Issue 4AFM Supplement AMFST06001 Issue 1</td>
<td>Form 1 Modification Statement LN/AM/706 Issue 4AFM Supplement AMFST06001 Issue 1</td>
<td>06/02/2008</td>
<td>Active</td>
<td>10014502</td>
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<td>EASA.A.S.02573</td>
<td>SABENA TECHNICS BGC</td>
<td>A300 B4 MODELS A300 B2 MODELS</td>
<td>M07 EGPWS Installation on A300 B2- B4Config 1: EGPWS with Dual FPI-910Config 2: EGPWS with Dual PPI-48Config 3: EGPWS with Dual PPI-4B and reactive Windshield</td>
<td>- Config 1: Master Data List n° 0121-01-A-2103, R0Aircraft Flight Manual Supplement ref. 0121-01-A-2301-R05 or later EASA approved revision Config 2: Master Data List n° 0121-02-A-2103, R0Aircraft Flight Manual Supplement ref. 0121-02-A-2301-R05 or later EASA approved revision Config 3: Master Dat...</td>
<td>Instructions for Continued Airworthiness are described in the Airworthiness Data Sheet Supplement AMFST05005 Issue 0121-01-A-2203-R05 or later EASA approved revision.</td>
<td>03/01/2006</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>LUFTHANSA TECHNIK AG</td>
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<td>BOEING 747</td>
<td>Installation of an ATC Mode S - Enhanced Surveillance Upgrade in Boeing 747 series aircraft</td>
<td></td>
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<td>EASA approved Airplane Flight Manual Supplement, Lufthansa Technik Document A-34-06/135-AMF, Rev IR or later EASA approved revisions required part of this STC</td>
<td>04/08/2006</td>
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<td>EASA.A.S.02576</td>
<td>SOCATA</td>
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<td>TBM 700</td>
<td>STC 34.001.06/Gamin GNS530 TAWS.</td>
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<td>TBM 700 equipped Garmin GNS 530 only s/n to 433 excepted 269.</td>
<td>30/01/2008</td>
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<td>EASA.A.S.02583</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>AIRBUS A310-300 AIRBUS A300-600</td>
<td>Installation of a new Stretcher for Kuwait Airways.</td>
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<td>15/11/2007</td>
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<td>EASA.A.S.02586</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>AIRBUS A320</td>
<td>Installation of a new Stretcher for Kuwait Airways</td>
<td></td>
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<td>28/02/2008</td>
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<td>10014510</td>
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<td>EASA.A.S.02588</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>BOEING 777-200 SERIES</td>
<td>Installation of a stretcher in the passenger compartment for Kuwait Airways</td>
<td></td>
<td></td>
<td></td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>06/11/2007</td>
<td>Active</td>
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<td>EASA.A.S.02589</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>BOEING 747-400</td>
<td>Installation of a new stretcher for Kuwait Airways</td>
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<td>06/11/2007</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>AIRBUS A340-300</td>
<td>Installation of a new Stretcher for Kuwait Airways</td>
<td></td>
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<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>LOCKHEED 188</td>
<td>Installation of Mode S Transponders for Enhanced Surveillance</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>B200</td>
<td>T AS620 Traffic Advisory System</td>
<td>FR Aviation Modification DC-0066FR Aviation AFM Supplement FRAH/FMII</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked...</td>
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<td>AERODATA AG</td>
<td>B300</td>
<td>Installation of Aerodata Flight Inspection System 4D-7FIS-0220.</td>
<td>Installation in accordance with Flight Manual Supplement AFM-082 Rev 0, or later approved revisions; Instructions for Continued Airworthiness AMM-816 Rev 0, or later approved revisions.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>PA-46-310T, PA-46-350P</td>
<td>Avionics upgrade with following equipment (DFE-DO-0075-EPL-01 Equipment and Parts List Rev. 1/1 GMA 347 Audiosystem Garmin 1 GNS 530 COM NAVI GPS, G1000 Y GNS 430 COM NAVI GPS Garmin G430</td>
<td>DFE work order package DFE-DO-0075 and Installation Manuals according to Reference Document list, DFE-DO-75-ECO-01 Rev 0, and 2-Approved Flight Manual Supplements DFE-DO-0075-AMS-430 REV IR, DFE-DO-0075-AMS-530 REV IR, DFE...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>BBJ DESIGN SERVICES LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>767-300ER</td>
<td>Installation of a MA5200E in Flight Entertainment System on Austrian Airlines B767-300ER aircraft</td>
<td>BBJ Design Services Modification Reference BB767MNO280-D70 Instructions for Continued Airworthiness NO280-D71 Maintenance Manual Supplement</td>
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<td>EASA.A.S.02603</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV 1</td>
<td>AIRBUS A300-600</td>
<td>A300-600 Cabin Reconfiguration from Current Continental Version to Intercntinental Layout</td>
<td>Compliance Record List CRL25-06/13M, Issue 2 dated 30 May 2006 Supplemental Type Definition Document STD25-06/13M, Issue 1 dated 8 June 2006</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>STC TWENTY ONE LIMITED</td>
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<td>CAA UK RESPONSIBIL</td>
<td>Boeing 727 SERIES</td>
<td>Introduction of Emergency Locator Transmitter</td>
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<td>Validation of definition specificities of TB200 S/N 1633 aircraft, compared to TB200 GT certified aircraft definition</td>
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<td>STC 53.001.06 applicable to TB200 s/n 1633 only</td>
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<td>Mode S Elementary &amp; Enhanced Surveillance System Installation</td>
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<td>JET AVIATION AG, BASEL</td>
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<td>B727-200</td>
<td>ADI HIS &amp; FMS upgrade by: removing both existing 3298-4x ADI 3295 removing both existing 3314-AD HSI removing both LTN 9202 removing both ALI 55B Radio Altimeter Indicators; installing two new S ATI E2; p/n 195000-2; installing two new ATI E1; p/n 195000-2; installing two new EFIS Control Panel; p/n;...</td>
<td>Jet Aviation Declaration of Compliance for the project BACB/B60 Master Data List, Doc. No. BACB/B60-002, latest approved revision, AFMS for the S ATI EFIS, Doc. No. BACB/B60-001-01, Rev. B or latest approved revision, AFMS for the CMC-800 FMS, Doc. No. BACB/B60-001-02, Rev. C or latest approved revision;</td>
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<td>FALCON 10</td>
<td>DFS 1274 std 5.0.6: Mode S Transponder Enhanced Surveillance</td>
<td>see Change approval sheet DFS 1274 AFM Supp F10 DFS 35 iso7TM 721(A) with DFS 1274 acno. &quot;Mode S Transponder Enhanced Surveillance&quot;, Wiring Diagram DFS10 600 34-56-01XX, DFS10 600 34-56-11XX (Optional), DFS10 600 34-42-03XX, DFS10 600 34-18-01XX and associated list of affected technical documents. Applicable to Falcon 10 with DFS 1269B applied previously or simultaneously</td>
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<td>CESSNA 404</td>
<td>Installation of a video under Cessna 404</td>
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<td>BOEING 737-500</td>
<td>B737-500</td>
<td>Installation of a video under Boeing 737-300/500/600 in accordance with the Modification File MF-B737-06-004, Rev 00, dated 11 October 2006. - Modification File MF-B737-06-004, Rev 00, dated 11 October 2006 - AFMS B737-06-004-001 First Issue dated 3 May 2006, or later approved revision. AFMS B737-06-004-002 First Issue dated 9 May 2008, or later approved revision. ICA B737-06-004 Rev 01, or later approved revision. 1. Limited to installation on Boeing 737-300/500/600. 2. Compatibility of this design change with other previously approved modifications must be determined by the installer.</td>
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<td>Limitations/Conditions</td>
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<td>ACS Change 7 Upgrade &amp; Elementary Surveillance Upgrade</td>
<td>Bouremouth Aviation (Consultants) Ltd BAC/DCH8000Bouremouth Aviation (Consultants) Ltd AFM Supplement FMS/38756 Issue 1 dated 15 June 2006 or later approved revisions</td>
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<td>BOMBERDAIR DHC-8-100 SERIES</td>
<td>Installation of HF System</td>
<td>Bouremouth Aviation (Consultants) Ltd BAC/DCH8000Bouremouth Aviation (Consultants) Ltd AFM Supplement FMS/38756 Issue 1 dated 15 June 2006 or later approved revisions</td>
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<td>EASA.A.S.02621</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
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<td>REV 1</td>
<td>BOMBERDAIR DHC-8-103 BOMBERDAIR DHC-8-102</td>
<td>Installation of Dual UNS-1E Flight Management System Note: Rev. 1 to STC EASA.A.S.02621 adds RNP Approach Capability.</td>
<td>- Bouremouth Aviation (Consultants) Ltd Doc BAC/DCH8501 Issue 6 or later approved revision - Bouremouth Aviation (Consultants) Ltd Flight Manual Supplement FMS/38753 Issue 4 or later approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer. 2. Prior to installation of this modification the installer...</td>
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<td>BOMBERDAIR DHC-8-100 SERIES</td>
<td>Installation of Aircell ST3100 Satellite Telephone System (SATCOM)</td>
<td>Bouremouth Aviation (Consultants) Ltd BAC/DCH8000Bouremouth Aviation (Consultants) Ltd AFM Supplement FMS/38756 Issue 1 dated 6 July 2006 or later approved revision</td>
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<td>BOMBERDAIR DHC-8-100 SERIES</td>
<td>Installation of Universal Terrain Awareness Warning System (TAWS)</td>
<td>Bouremouth Aviation (Consultants) Ltd BAC/DCH8000Bouremouth Aviation (Consultants) Ltd AFM Supplement FMS/38756 Issue 1 dated 6 July 2006 or later approved revision</td>
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<td>MITSUBISHI MU-300</td>
<td>Installation of Universal ULS-1L(-1K(+)) FMS</td>
<td>Engineering Order 20001552-EO-01 Issue 2, dated 29.04.2006 or later approved Revisions Aircraft Flight Manual Supplement 2001552-AM-01 Issue 2 dated April 2006 or later approved Revisions</td>
<td>Certification Basis of the Change: CS 25 Amendment 1, additionally JAA Leaflet No. 10 (Precision (RNAV), AMC 20-4 (Basic(RNAV)), AMC 20-5 (GPS))</td>
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<td>GULFSTREAM GV-SP</td>
<td>Certification of FAA Field Approved Modifications to enable the issue of a Certificate of Amortenousness on the Swiss register. Aircraft Serial No 5040</td>
<td>In accordance with Gulfstream Aerospace Limited Change Leaflet for Change No GAeL 06102 Issue 2</td>
<td>Aircraft Serial No 5040 Only</td>
<td>24/08/2006</td>
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List of EASA Supplemental Type Certificates

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<td>EASA.A.S.02634</td>
<td>AVONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>PA-31P</td>
<td>Installation of COM/NAV/GPS Garmin GNS530/GNS430 (dual installation)</td>
<td>EASA Approved Flight Manual Supplement ASR 2006-025-00 Revision A or later EASA approved revision</td>
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<td>Cockpit Voice Recorder CVR-308 Installation (Universal Avionics System Corp.)</td>
<td>Engineering Order, 220-LR, Rev. IR, or later approved revision/AFM Supplement, AG581-071, Rev. IR, or later approved revision</td>
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<td>Solid State Flight Data Recorder F1000 Installation (L-3-AR)</td>
<td>Engineering Order, EO 219-LR, Rev. IR, or later approved revision/AFM Supplement AG583-071, Rev. IR, or later approved revision</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>CESSNA 551, Cessna Citation I model 501 and Cessna Citation I model 551</td>
<td>Installation of Dual Garmin GNS 430A/530A and 3rd COMM in Cessna Citation I model 501 and Citation II model 551.</td>
<td>• Engineering Order 2027496-EO-01, issue 3, dated 30 May 2007, or later approved revision. • AFM Supplement Doc. 2027496-01A, issue 2, dated 31 May 2007, or later approved revision.</td>
<td>1. Certification basis of the Change CS-23, additionally: 2. AMC 20-4 (Basic RNAV), AMC 20-5 (GPS), AMC 25-11 (Electronic Display Systems) and JAA TGL 7, rev.1 (VHF Com 8.33 KHz) 3. Prior to installation of this modification the installer must determine that the interrelationship between the...</td>
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<td>CESSNA 550, Cessna Citation I model 500 and Cessna Citation I model 550,</td>
<td>Installation of Dual Garmin GNS 430A/530A and 3rd COMM in Cessna Citation I model 500 and Citation II model 550,</td>
<td>• Engineering Order 2027496-EO-01, issue 3, dated 30 May 2007, or later approved revision. • AFM Supplement Doc. 2027496-01A, issue 2, dated 31 May 2007, or later approved revision.</td>
<td>1. Certification Basis of the Change CS-25 Amdt. 2, additionally: 2. AMC 25-4 (Basic RNAV), AMC 20-5 (GPS), AMC 25-11 (Electronic Display Systems) and JAA TGL 7, rev.1 (VHF Com 8.33 KHz) 3. Prior to installation of this modification the installer must determine that the interrelationship between the...</td>
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<td>SD-60 SHERPA SD-33; SD-60; SD-3 SHERPA</td>
<td>Honeywell EGPS Mk VI Installation</td>
<td>Engineering Order 2032334-E0, Issue 2 dated 15 May 2006 or later approved revision</td>
<td>EASA STC no. 10014559 Certification Basis of the Change CS 25, additionally JAA TGL 12 (TAWS)</td>
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<td>SD-60 SHERPA SD-33; 3-60, 3-SHERPA, 360 SH</td>
<td>Trimble TML-2101 I/O Approach + GPS System Installation</td>
<td>Engineering Order 2032335-E0, Issue 2 dated 15 May 2006 or later approved revision</td>
<td>Certification Basis of the Change CS 25, additionally EASA AMC 20-4 (BRNAV) and AMC 20-5 (GPS)</td>
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<td>FMS UNS-1L Installation (Universal Avionics System Corp.)</td>
<td>Engineering Order, 218-LR, Rev. IR, or later approved revision</td>
<td>AGS86-071, Rev. A, or later approved revision</td>
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<td>IBERIA LINEAS AEROGAS DE ESPANA</td>
<td>AIRBUS A321-212</td>
<td>Passenger Cabin Lay-Out Modification for Special Flight</td>
<td>Master Data List MOD-MDL-018-01 dated 6 December 2006 or later approved revision</td>
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<td>EASA.A.S.02662</td>
<td>TELEDYNE Ltd/D BIA TELEDYNE CONTROLS FIGHT DATA COMPANY</td>
<td>MB8 SERIES</td>
<td>Installation of Optical Quick Access Recorder</td>
<td>Master Data List MOD-MDL-018-01 dated 6 December 2006 or later approved revision</td>
<td>05/07/2007</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOMBARDIER CL-600-2B16 (604)</td>
<td>STC Twenty One MDL No. S21-MDL-0131 at Issue 1 or later approved revision. Midcoast Aviation Inc. AFM supplement 34-261M021 or later approved revision.</td>
<td>22/06/2006</td>
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<td>SCHERDEL ENGINEERING Ltd.</td>
<td>LS1-F</td>
<td>Change of Empty Mass and Ballast Requirements for the Glider LS1-F</td>
<td>Doc Number 06-262-001</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation&quot; Compatibility with other aircraft configurations shall be determined by the installer.</td>
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<td>EASA.A.S.02668</td>
<td>BJL DESIGN SERVICES LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BD-700-1A11</td>
<td>Installation of an Electronic Flight Bag (EFB) on A/C IN 9663 Modification Summary List: BJL7/45 Issue 1CCL. BJUCCL09 Issue 1A&amp;DC. Mod. Sum. Sh. A Certificate of Design iss. 25/10/2006</td>
<td>AFM-Supplement: BJL/JM00717 Rev 19Installation Manual 929-60040-00 Rev 4</td>
<td>This approval of this installation does not include any approval of the suitability, integrity, or accuracy of the EFB hardware, software, data base, display, controls or lighting. The capability of the system to &quot;perform its intended function&quot; has not been verified or certified. This approval does no...</td>
<td>03/11/2006</td>
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List of EASA Supplemental Type Certificates

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<td>EASA.A.S.02673</td>
<td>KUERZI AVIATION AG</td>
<td>BOMBARDIER DHC-6 SERIES 300</td>
<td>Installation of KMH-880, GPS-400 &amp; KMD-850.</td>
<td>MDG.200301247 Rev. C or later approved revisions.</td>
<td>Limited to S/N 0841.</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>BOEING 767</td>
<td>B767 Cabin Reconfiguration from KLM to Aerosol.</td>
<td>CAS-8767-06-008</td>
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<td>EASA.A.S.02675</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>CITATION CS60XLS</td>
<td>Installation of Honeywell Solid-State Flight Data Recorder (SSFDR) with/without Controls Mini Flight Data Acquisition Unit (MFDAU)</td>
<td>Engineering Order, 036A-EO, dated 06/06/2004 or later approved revision</td>
<td>AFM Supplement, 036A-AMS, dated 10/05/2006 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configurations shall be determined by the installer.</td>
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<td>EASA.A.S.02676</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CORPORATION AG</td>
<td>HAWKER 800 SERIES BAE 125 -800</td>
<td>Replacement of Dual FMS Units-1D by Dual FMS-1F.</td>
<td>Installation to be in accordance with AEC Engineering Order ref: A569-062, rev A, dated 05.09.2006 or later approved revision.</td>
<td>Operation to be in accordance with AEC Aircraft Flight Manual Supplement ref. A569-071, rev. R, dated 14.06.2006 or later approved revision. Continued Airworthiness to be in a...</td>
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<td>SABENA TECHNICS DINR</td>
<td>FOKKER F.28 MARK 0100</td>
<td>Fokker 100 VIP Cabin</td>
<td>AFM Supplement FK-2S-LAIA-01-PM REV H</td>
<td>Maximum approved passenger capacity 19. Limited to serial numbers 11368, 11403, 11406. Refer to AFM Supplement.</td>
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<td>EASA.A.S.02679</td>
<td>SELL GmbH</td>
<td>BOEING 777-300 BOEING 777-200</td>
<td>Interior Re-configuration of Forward Galley Complex for Emirates airline</td>
<td>Type Investigation Document MD10332-25-01 iss. 1 dated January 17, 2007.</td>
<td>Limited to B777-200-300 models serial numbers 79324 and 79325.</td>
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<td>EASA.A.S.02683</td>
<td>AIR ALPHA A/S</td>
<td>B200</td>
<td>Installation of Trane &amp; Trane Avio-1 satcom equipment in accordance with danish STC 04998.Note: This STC superseded SLV STC No. 3/98</td>
<td>Copenhagen Avoritice T.O package: &quot;I &amp; T Aero-I Satcom installation&quot; AFM Supplement. Trane &amp; Trane Avio-I Satcom system in Baech model B200 Danish STC 03/09.</td>
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<td>TENENCIA Ltd.</td>
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<td>EASA.A.S.02689</td>
<td>BAE 146/AVRO 146</td>
<td>Installation of Single or Dual Universal UNS-IL or UNS-RK Flight Management System</td>
<td>Tenencia Modification TEN 4674AFM Supplement 1 issue 1 (TEN/146/001)AFM Supplement 2 issue 1 (TEN/146/002)AFM Supplement3 issue 1 (TEN/146/003)AFM Supplement 4 issue 1 (TEN/146/004)</td>
<td>RNAV Approaches are not permitted for installation with electronic-mechanical instruments</td>
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<td>REV. 1</td>
<td>EASA.A.S.02691</td>
<td>MYSTÈRE-FALCON 20C/DS/ES</td>
<td>Installation of new avionics: Pylon/Phoenix/367/02086</td>
<td>see Change Approval Sheet CAS AVD MDA20-95-054B Issue 1 dated 10/08/2006</td>
<td>1. Limitations: no adverse effect on the airworthiness of the aircraft 2. Prior to installation of this modification the installer must determine that the compatibility and performance between this modification and any other previously installed modification will introduce no adverse effect upon the aircrafts...</td>
<td>24/11/2008</td>
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<td>EASA.A.S.02692</td>
<td>MYSTÈRE-FALCON 20C/DS/ES</td>
<td>Installation of under-wing stores</td>
<td>see Change Approval Sheet CAS AVD MDA20-95-055A Issue 1 dated 10/08/2006</td>
<td>1. Limitations: no adverse effect on the airworthiness of the aircraft 2. Prior to installation of this modification the installer must determine that the compatibility and performance between this modification and any other previously installed modification will introduce no adverse effect upon the aircrafts...</td>
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<td>MYSTÈRE-FALCON 20C/DS/ES</td>
<td>STRUCTURAL INSTALLATION OF A FUSELAGE FUGRO GPS ANTENNA</td>
<td>Change Approval Sheet N CAS AVD MDA20-34-052A Issue 1 dated 10/08/2006</td>
<td>Compatibility with previous approved modifications shall be determined by the installer.</td>
<td>23/04/2007</td>
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<td>EASA.A.S.02694</td>
<td>MYSTÈRE FALCON 20</td>
<td>Installation of cones in cabin.</td>
<td>see Change Approval Sheet CAS AVD MDA20-25-056A Issue 1 dated 10/08/2006</td>
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<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BOEING 737-200</td>
<td>Installation of 108 Seat LOPA</td>
<td>Bouremouth Aviation (Consultants) Modification BAC/737/056</td>
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<td>PHOENIX AEROSPACE LIMITED</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BN2B-26 ISLANDER</td>
<td>Installation of GNS530, GNS430 &amp; GMA340.</td>
<td>Phoenix Aviation Design Services Modification PADS/CP/1057-Phoenix Aviation Design Services AFM Supplement PADS/FMS/B2B/001 Revision 0 or later approved revision</td>
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<td>ARTOURING LTD</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>PA-31-350 (CHIEFTAIN)</td>
<td>Installation of Avionics Upgrade</td>
<td>Artouring Modification AT MOD_1096 Avionics Upgrade AT FMS_1096</td>
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<td>AMCALE DE VOLTIGE AERIENNE(AVA)</td>
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<td>CAP231, CAP231EX, CAP232</td>
<td>Fuselage frames 1,2 &amp; 3 Reinforcement</td>
<td>-SB AVA-B5-2006-02 for CAP231 aircraft and Fuselage Frames 2 &amp; 3 for CAP231EX aircraft</td>
<td>1. Fuselage Frames 2 &amp; 3 for CAP232 aircraft and Fuselage Frames 2 &amp; 3 for CAP231EX aircraft must be reinforced by Art Menuiserie only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with this STC is approved only...</td>
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<td>EASA.A.S.02710</td>
<td>FR AVIATION LTD/DIA COBHAM AVIATION SERVICES UK</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>MOD/DF/963 FAN JET &amp; MYSTERE FALCON 20012 APPLICABLE TO FALCON 20 BASIC,</td>
<td>DC-1075 Issue 2:bl: 8 Mar 07 - Enhanced ATC Mode S or later approved revision: Enhanced/AT Mode S.</td>
<td>See Design Change Statement DC-0075:bl: 8 Mar 07 (includes Master Datalist): Flight Manual Supplement FRM/DF/963 Rev. 7:bl: 27 Feb 07 or later approved revision. Instruc ations for Continued Airworthiness IAC-0075 Rev. A or later approved revision</td>
<td>1. Certification Basis CS25 initial issue 2. Compliance with TOL 13 and NP A 20-12a has been shown 3. Applicable to Falcon 20 Basic, D &amp; E models Post Mod FDB036.4. Not all required Download Parameters/ DAP are provided, the AFM/Limitation Section contains the DAPs available 5. This STC is approved only...</td>
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<td>EASA.A.S.02711</td>
<td>EXECUTIVE JET GROUP LTD</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>900</td>
<td>Approval of Cabin Configuration &amp; Changes for the Issue of an EASA C d/cA on Aircraft Reg/G-HMEV</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with otherwise aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered overw...</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BAE ATP</td>
<td>Rework of LH Passenger Door</td>
<td>STC Twenty One Master Drawing List 521 MDL 0135 Initial issue dated 16 April 2007 or later approved revision STC Twenty One AFM Supplement 521 FMS-0016 Issue 1 or later approved revision.</td>
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<td>AERO TECHNICS DESIGN LIMITED</td>
<td>REV 1</td>
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<td>BOEING MD-83</td>
<td>Cabin Reconfiguration from 16Y to 52C/22Y Seating</td>
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<td>BOEING 767-300</td>
<td>Enhanced Surveillance Dual ATC Mode S Installation</td>
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<td>EASA.A.S.02728</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>CESSNA 650 CITATION VII CESSNA 650 CITATION III</td>
<td>Installation of Honeywell Multi-Function Radar Display (MFDR) and as an option a Honeywell Primus 400 Radar system</td>
<td>Installation in accordance with Scandinavian Avionics Engineering Order ref 2034325-EC Issue 2, dated 28 June 2006, or later approved revisions.Operation in accordance with Scandinavian Avionics Flight Manual Supplement ref 2034304-AM Issue 1, dated August 2006, or later approved revisions.Instructions...</td>
<td>This STC is only applicable to Aircraft without EFIS system and integrated MFD.</td>
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<td>EASA.A.S.02730</td>
<td>LUFTHANSA TECHNIK AG</td>
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<td>BOEING MD 11</td>
<td>Installation of a Rockwell Collins Tailwind 550 Live TV system in Boeing MD-11 as detailed in STDD5-06/19M Supplemental Type Definition Document for Tailwind 550 Live TV installation MD-11</td>
<td>1. A-00-07/10/AFM Aircraft Flight Manual Supplement for TW550 Live TV installation, Rev. 01 dated November 06, 20072. R-00-07/03/ICA Instructions for Continued Airworthiness</td>
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<td>AERODISA S.L.</td>
<td>Active</td>
<td>EMB-120, EMB-120ER, EMB-120RT</td>
<td>AED2006-04M</td>
<td>TAWS Installation on EMBRAER 120</td>
<td>• Master Data List AED2006-04M_MDL. Revision 6, dated 30/04/2007 or later EASA approved revision.</td>
<td>1. See FM Supplement AED2006-04M_MDL_D03 Section II: LIMITATIONS.2 Certification basis per ANAC TCDS EA-855.</td>
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<td>AERODISA S.L.</td>
<td>Active</td>
<td>SA227-PC, SA227-AC, SA227-8C</td>
<td>AED2006-04M</td>
<td>TAWS installation on Fanchti SA 220/227</td>
<td>Master Data List AED2006-04M_MDL. Revision 3, dated 19/04/2007 or later EASA approved revision</td>
<td>See FM Supplement AED2006-04M_MDL_D03 Section II: Certification basis per FAA TCOG A550: FAR 23 Amendment 8.</td>
<td>03/05/2007</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>Active</td>
<td>CESSNA 525</td>
<td>Installation of KLN-90B Model: 525 S/N 0001 up to 0359</td>
<td>Engineering Order 119-, EORUAG AFM-5 118-PM, Rev. 0, dated 10. March 2004, or later EASA approved revision</td>
<td>S/N 525/0001 up to 525-0359 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>Active</td>
<td>AIRBUS A340-300</td>
<td>THY A340 Cabin Retrofit Modification</td>
<td>Air France Change Approval Sheet 0 CAS A340-36-007 Rev. 0 Edition 0</td>
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<td>EASA.A.S.02741</td>
<td>Aeroservis s.r.o.</td>
<td>Active</td>
<td>CESSNA T210 SERIES</td>
<td>Upgrade of NAV/COM equipment for IFR navigation.</td>
<td>Pilot's operating handbook supplement No. 40, 41, 42, 43, 44, 45, 46, 47.</td>
<td>1. S/N 210638072. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
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<td>EASA.A.S.02745</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>Active</td>
<td>BEECH 1900D</td>
<td>Installation of VIP configuration in aft ward cabin and passive noise reduction kit</td>
<td>Airplane Flight Manual Supplement, Rev. 0, dated 13 July 2006 or later EASA approved revision.</td>
<td>This modification limits 14 passengers maximum.</td>
<td>08/11/2006</td>
<td>Active</td>
<td>1014615</td>
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<td>EASA.A.S.02747</td>
<td>DASSAULT FALCON SERVICE</td>
<td>Active</td>
<td>MYSTERE FALCON 20-5 MYSTERE FALCON 200 MYSTERE FALCON 200 FAN JET</td>
<td>DFS536 issue A.51, 29 Sep’06</td>
<td>see Master Document List DFS-GE-17-F3-536 Issue A1 dated 29 Sep’06 or later EASA approved revisions</td>
<td>Leaflet TGL 12 “Certification Considerations for the TerrainAwareness Warning System: TAWS” applies</td>
<td>02/10/2006</td>
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<td>EASA STC no.</td>
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<td>Associated Technical Documentation Limitations/Conditions</td>
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<td>EASA.A.S.02751</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
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<td>FML-200</td>
<td>- Installation of 5-blade MT-27-1L-E-C- F-RG(2) / CFR7205-S foil Full Feathering and Reversing Propellers Revision 2. Extension of STC EASA.A.S.02755 to airplane model Do-228-212 Revision 2 and 3 issued to correct some typos. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>DFS2374: Limitation Removal</td>
<td>Applicable to Fan Jet Falcon Sr N 328 only</td>
<td>03/06/2006</td>
<td>Active</td>
<td>10114173</td>
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<td>EASA.A.S.02757</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
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<td></td>
<td>FML-200</td>
<td>Honeywell Primus II RZ2-833 () Integrated Radio System – EHS.</td>
<td>RUAG Aerospace Services Master Drawing List 179E-MDL.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>12/01/2007</td>
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<td>EASA.A.S.02758</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
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<td>FML-200</td>
<td>Upgrade Dual Honeywell FMS NZ-2000 / PRNAV.</td>
<td>RUAG Aerospace Services Master Drawing List 179B-MDL.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>12/01/2007</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
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<td>FML-200</td>
<td>Iridium Satcom ST3100.</td>
<td>RUAG Aerospace Services Master Drawing List 179C-MDL.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>12/01/2007</td>
<td>Active</td>
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<td>EASA.A.S.02760</td>
<td>AERODATA AG</td>
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<td></td>
<td>FML-200</td>
<td>Installation of TCAS 2000 System including Mode S Transponders</td>
<td>MDL-AC-274 Rev.5 Master Document ListAFM - 08 Rev. 2 Airplane Flight Manual Supplement/Ref-A/ C-747 Maintenance Manual Supplement</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked...</td>
<td>20/07/2006</td>
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<td>10114300</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.02764</td>
<td>GAMA ENGINEERING LTD.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>HAWKER 800 XP</td>
<td>Installation of a L-3 Avionics F1000 Flight Data Recorder.</td>
<td>Certification basis</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>17/10/2006</td>
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<td>10014625</td>
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<td>GAMA ENGINEERING Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BEECH 200</td>
<td>Enhanced Mode S Upgrade.</td>
<td>Master Document List</td>
<td>This STC however grants no operational approval to this installation, that will need to be obtained from the States for the particular aircraft involved as defined in JAR-OPS 1.715(g).</td>
<td></td>
<td>12/03/2008</td>
<td>Active</td>
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<td>EASA.A.S.02768</td>
<td>AIM AVIATION (JECCO) LTD.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>A330-200/A330-300 MODELS</td>
<td>Introduction of flight crew rest facility into A330-200 and 300 aircraft.</td>
<td>Major change kW 334/36-SB 3AM330-25-004</td>
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<td>31/05/2007</td>
<td>Active</td>
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<td>EASA.A.S.02773</td>
<td>AEROTECHNIC GMBH</td>
<td>DORNIER DO 228-212</td>
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<td>Installation of mission equipment for maritime surveillance.</td>
<td>Restricted to S/N R181, 8133.</td>
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<td>26/07/2007</td>
<td>Active</td>
<td>10014632</td>
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<td>EASA.A.S.02777</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td></td>
<td>55</td>
<td>Installation of TAWS / RMI ST3400 (Sandel) System</td>
<td>Certification Program E/O 221-LR, rev. 2AFM Supplement A5601-071, rev. A</td>
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<td>23/01/2007</td>
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<td>EASA.A.S.02782</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>AIRBUS A330</td>
<td>Installation of Quick Access Recorder.</td>
<td>Compliance Document Summary Sheet and Accomplishment Instructions Continuation Sheet, Ref BACABUS/0009, Issue 2 dated 19 December 2006.</td>
<td></td>
<td>02/01/2007</td>
<td>Active</td>
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<td>Limitations/Conditions</td>
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<td>EASA.A.S.02783</td>
<td>EAD AEROSPACE</td>
<td>REV 1</td>
<td></td>
<td>BOEING 767</td>
<td>Installation of Panasonic “EFX” Passenger in Flight Entertainment System (IFE).</td>
<td>EAD Aerospace Master Drawing List 0110-EAS-A-MDL at Revision G-Airplane Flight Manual Supplement 0110-EAS-A-AFM-01 at Revision C.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs “Description” and “Associated Technical Documentation”. Compatibility with other aircraft configuration shall be determined by the installer.</td>
<td>04/04/2007</td>
<td>Active</td>
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<td>EASA.A.S.02784</td>
<td>EAD AEROSPACE</td>
<td></td>
<td></td>
<td>BOEING 737-900 BOEING 737-800 BOEING 737-700 BOEING 737-600</td>
<td>Implementation of Honeywell “Link 2000” Controller-Pilot Data Link Communications (CPDLC) on B 737</td>
<td>- Supplemental Type Design Definition Doc 01106-EAS-A-STD-01 Revision C, or later EASA approved Revision; - Master Drawing List 01106-EAS-A-MDL, Revision E or later EASA approved Revision; - AFM Supplement 01106-EAS-A-AFM-01 Revision B, or later EASA approved revision; - MEL Supplement 01106...</td>
<td>Compatibility of this design change with other previously approved modifications must be determined by the installer.</td>
<td>09/03/2007</td>
<td>Active</td>
<td>10014330</td>
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<td>EASA.A.S.02785</td>
<td>AVIATION DEFENSE SERVICE</td>
<td></td>
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<td>MYSTERE FALCON 20</td>
<td>AVD-MDA20-49-053A: Additional Power Bus using APU generation, installation of 2 additional inverters.</td>
<td>see Change Approval Sheet CAS AVD-MDA20-49-053A Iss. 0 dtd. 18.4.06 or later approved revision - AFM supplement to DTM 918 - AFMS FA20 rev 12 dtd. 10.12.07 or later approved revision and associated technical documentation</td>
<td>Limitations iaw AFM supplement to DTM 918 - AFMS FA20 apply for operation with additional inverters installed iaw AVD-MDA20-49-053A.</td>
<td>10/01/2008</td>
<td>Active</td>
<td>10014339</td>
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<td>EASA.A.S.02789</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td></td>
<td></td>
<td>AEROSPATIALE ATR 72 AEROSPATIALE ATR 42</td>
<td>Indeavor Siren Cargo Container System installation on ATR42-320 and ATR72-202 aircraft previously converted to Class E Cargo Compartment Freighter.</td>
<td>For installation on ATR42-320 aircraft Bournemouth Aviation/BAC/ATR/003 Issue 2 Siren Cargo Loading System Installation ATR42-320 and Installation Instructions as per Bournemouth Aviation Accomplishment Instructions BAC/ATR/003 Issue 2 and Drawing BAC-1522-01 Issue B • Maintenance Instructions as...</td>
<td>This STC is applicable on the following aircraft: Those ATR42-320 previously converted to Class E Cargo Compartment Freighter by the embodiment of the following ATR GIE Service Bulletins: • ATR42-25-0070 Reinforce all floor panels, and • ATR42-25-0063 Modify A/C for Cargo Container Loading, and...</td>
<td>29/01/2008</td>
<td>Active</td>
<td>10014340</td>
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<td>EASA.A.S.02791</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
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<td>CESSNA 560 CESSNA 550 BRAVO</td>
<td>Honeywell EGW5 MK VIII Installation</td>
<td>Engineering Order 2035111-E0 Issue 1, dated 16/08/2006 or after approved service/aircraft Flight Manual Supplement 2035111-AM Issue 1, dated August 2006 or later approved revision</td>
<td>Certification Basis of the Change: CS 25 Amendment 1m, additionally JAA Leaflet No. 12 (TAWIS)</td>
<td>03/11/2006</td>
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<td>EASA.A.S.02792</td>
<td>TSA TRANSAIRCO S.A.</td>
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<td>PILATUS PC-12/47 PILATUS PC-12/45 PILATUS PC-12</td>
<td>AFT Refreshment Galley and Oven Refer to master certification documents list TS PC12-25.30.02MCL</td>
<td>Airplane Flight Manual Supplement TSA PC-12-25.30.02AFM8</td>
<td>1. Limited to serial number 321 and later.2. Galley power and microwave oven is not approved during take-off and landing phases of flight and during emergency or abnormal procedures.3. When aft refreshment galley is installed, the luggage area maximum load is 218lb (100Kg) and limited 1...</td>
<td>21/11/2006</td>
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<td>EASA.A.S.02794</td>
<td>FR AVIATION LTD/D/BA COBIHAM AVIATION SERVICES UK</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td>FAN JET FALCON BASIC, D AND E</td>
<td>DC-9/04 Issue 1: Extended RVSM operation</td>
<td></td>
<td>1. DC-9/04 Issue 1 dated 11.12.07 or later approved revision.2. FRA AFM Supplement FRA/HFM/00 revision 9 or later approved revision.</td>
<td>1. Limited to Fan Jet Falcon with FRA STC EASA.A.S.0050 embodied.3. This STC does not constitute an operational approval. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will not...</td>
<td>03/02/2008</td>
<td>Active</td>
<td>10101315</td>
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<td>EASA.A.S.02795</td>
<td>E.I.S.A.</td>
<td></td>
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<td>MYSTERE FALCON 20</td>
<td>Addition of Enhanced Surveillance System on existing TDR 94 Mode S Transponder installation.</td>
<td>Supplemental Type Certificate Data List EASA 05 SDL S114 Rev. 4Aircraft Flight Manual Supplement EASA 05 AFM S114 Rev. 1Aircraft Maintenance Manual Supplement EASA 05 AFM S114 Rev. 10 later approved revisions of these documents.</td>
<td>Major Design Change applicable to aircraft Fan Jet Falcon and Falcon 20 equipped with: • Rockwell Collins Proline IV avionics suite • Mode S Elementary Transponder TDR-94D-008 • FMS-6000 • Air Data Computer ADC-850C • An Input/Output Concentrator Aircraft Flight Manual Supplement EISAA...</td>
<td>01/02/2008</td>
<td>Orphaned</td>
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<td>EASA.A.S.02796</td>
<td>JET AVIATION AG, BASEL</td>
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<td>AIRBUS A319-100</td>
<td>Installation Of Additional Water Tank</td>
<td>Declaration of Compliance, Document Number BVCC/003-001, Revision A, dated 16th April 2007, or later EASA approved revision.</td>
<td>RUAG Aerospace GmbH Master Drawing List 719F-MDL.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>24/04/2007</td>
<td>Active</td>
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<td>EASA.A.S.02799</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
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<td>BOMBARDIER CL-600-2B16 (CL-601)</td>
<td>40Ah Battery with Charger.</td>
<td></td>
<td>RUAG Aerospace GmbH Master Drawing List 179F-MDL.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>12/01/2007</td>
<td>Active</td>
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<td>EASA.A.S.02800</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
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<td></td>
<td>BOMBARDIER CL-600-2B16 (601-3A)</td>
<td>Interior, Ancillar Antenna &amp; Miscellaneous Equipment.</td>
<td></td>
<td>RUAG Aerospace Services Master Drawing List 179G-MDL.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>12/01/2007</td>
<td>Active</td>
<td>10116446</td>
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<td>EASA.A.S.02803</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
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<td>CESSNA 560 (CITATION V)</td>
<td>Transponder Mode S Honeywell Primus I – ELS/ESHS, Upgrade of the existing Mode S Transponder RCZ-833K by Honeywell Service Bulletin to implement ELS/ESHS transponder capability. Installation of new wires from the FMS and the two Display/Guidance Computer IC-600 to the dual RCZ-833K Com Units EAS.</td>
<td>RUAG Compliance Check List Doc.-No:148C-CC1 Rev N/C dated 24.08.06, RUAG System Description Doc.-No:148C-SD Rev. N/C dated 18.08.PATS Aircraft, LLC/JS/AFM Supplement Doc.-No 148C- AMFS</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>15/09/2006</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
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<td>MOONEY M20M</td>
<td>BRNAV capacity modification: Installation of annunciator MD 41, technical description in F&amp;M No. 1147/06/SEAM</td>
<td>Installation of annunciator MD 41, technical description in F&amp;M No. 1147/06/SEAM</td>
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<td>Modification limited to s/n 27-0135</td>
<td>16/03/2007</td>
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<td>EASA.A.S.02810</td>
<td>ACI SYSTEMES</td>
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<td>BEECH 200</td>
<td>ATC replacement and Mode S Installation</td>
<td>ACI S007/2006/AFMS No 015 or later approved revision</td>
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<td>AFMS must be on board when modified</td>
<td>07/05/2007</td>
<td>Orphaned</td>
<td>10014653</td>
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<td>EASA.A.S.02812</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
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<td>CESSNA 650 CITATION VII</td>
<td>Installation of Honeywell EGWPS Mk VIII.</td>
<td>Installation in accordance with RUAG Master Document List Document No 178B-MDL and Engineering Order Document No 178B-EO, at the latest approved revision.</td>
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<td>None</td>
<td>11/12/2006</td>
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<td>EASA.A.S.02815</td>
<td>BBJ DESIGN SERVICES LIMITED</td>
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<td>CAA UK RESPONSIBILITY 747-400</td>
<td>Interior changes required on Cathay Pacific Boeing 747-400 aircraft MSN 27132 for BE Aerospace Flight Structures.</td>
<td>MOD BBJ 804</td>
<td></td>
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<td>10/05/2007</td>
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<td>EASA.A.S.02819</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>REV 1</td>
<td></td>
<td>CESSNA 650 CITATION VI</td>
<td>Enhanced Surveillance Mode S Transponder Upgrade (Honeywell)</td>
<td>EO 222-UL, rev.</td>
<td>1Certification Compliance Sheet AG599-075, rev. 5AAMF Supplement AG599-071, rev. 1</td>
<td></td>
<td>28/03/2007</td>
<td>Active</td>
<td>10116622</td>
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<td>EASA.A.S.02820</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV 1</td>
<td></td>
<td>BOEING 737-500</td>
<td>Installation of elementary surveillance Dual ATC Mode S on B737-300-500 in accordance with the Modification File MDF-B737-04-004 Rev 01, dated 11th October 2006.</td>
<td>- Modification Design File MDF-B737-04-004, Rev 01, dated 11th October 2006. - Modification Airmorethness File MAT-B737-04-004 Rev 01, or later approved revisions - AFBS 8737-04-004-34-002, first issue - ICA-B737-04-004 Rev 01, or later approved revision.</td>
<td>1. Limited to installation on Boeing 737-300-500. 2. Compatibility of this design change with other previously approved modifications must be determined by the installer.</td>
<td>23/10/2006</td>
<td>Active</td>
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<td>EASA.A.S.02823</td>
<td>DASSAULT FALCON SERVICE</td>
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<td>FALCON 10</td>
<td>MODE S TRANSPONDER WITH ENHANCED SURVEILLANCE CAPACITY</td>
<td>Saw:Master Document List DFS-DE-17-F1-1278 Issue B 1 or later approved revisions Aircraft Flight Manual Supplement DFS NP38, Rev Original or later approved revisions</td>
<td>1. This STC is approved only for those Falcon 10 aircraft equipped with two Collins TDR-94D-007 Mode “S” ATC Transponders and “Flight Ident” source 2. Compatibility with previously approved modifications shall be determined by the installer. 3. JAA Leaflet TGL 13 Certification of Mode S Transponder.</td>
<td>05/12/2006</td>
<td>Active</td>
<td>10116666</td>
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<td>EASA.A.S.02824</td>
<td>STC TWENTY ONE LIMITED</td>
<td></td>
<td></td>
<td>BOEING 757-200 SERIES</td>
<td>Reduction of Over-Wing Emergency Exits from 4 to 2.</td>
<td>AFM Supplement No. S21-FMS.0021.</td>
<td></td>
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<td>22/12/2006</td>
<td>Active</td>
<td>10116657</td>
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<td>EASA.A.S.02826</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td></td>
<td>L-410 UX/P-E20</td>
<td>Installation of Honeywell Enhanced Ground Proximity Warning System EGPWS/AWS.</td>
<td>+ AFM Supplement Doc. 2035430-AM, Issue 2, or later ESAApproved Issue+ Description of Change: 2035430-DE, Issue1, Engineering Order: 2035430-E0, Issue 2, Technical Order: 2035430-T0, Issue 1</td>
<td>1. This STC is only applicable for aircraft which have all the following equipment already installed:Honeywell (Bendix-King) IN-182 or KMD-850 Radar Display KN-53 NAV SystemKRA-405 Radar AltimeterKCS-55A Compass SystemComputer Instruments Corp. 04077 Air Data System2. This STC is</td>
<td>30/01/2007</td>
<td>Active</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.02627</td>
<td>FR AVIATION LTD/D/BA COBHAM AVIATION SERVICES UK</td>
<td>CA UK RESPONSIBILITY</td>
<td>MYSTERE FALCON 200/20 SERIES</td>
<td>FAN JET</td>
<td>DC-0088 Rev. 0.012.22 12.06: Increase to Max Zero Fuel Weight.</td>
<td>DC-0088 issued 22.12.06 or later approved revision- FRA AFM Supplement FRAHFM000 revision 6 or later approved revision</td>
<td>1. Limited to Falcon 20 acc. DC-0088 Master Document List Section 2, Cert Basis CAR 4b, and (Elast to Comply to) CS25 Amnd 2 paragraphs 25.1, 25.25, 25.57(14)(g)(25.1501(b), 25.1519, 25.1518 and 25.1583.3. The STC is approved only for the product configuration as defined in the approved design guideline r...</td>
<td>09/02/2007</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.02628</td>
<td>AERO-DIENST OMB &amp; CO KG</td>
<td>REV. 1</td>
<td>HS.125 SERIES 700B</td>
<td>FMS UNS-1L System Installation (AACS)</td>
<td>Installation in accordance with Aero-Dienst Engineering Order ref 223- HV, Rev. 1, dated 21, June 2007, or later approved revision. To be operated in accordance with Aero-Dienst Flight Manual Supplement ref AG83-071, Rev A or later approved revision</td>
<td>None</td>
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<td>07/01/2008</td>
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<td>EASA.A.S.02631</td>
<td>TRANSAERO S.A.</td>
<td></td>
<td>PILATUS PC-1247 PILATUS PC-1245 PILATUS PC-12</td>
<td>F.D.S. FD932 DVD Player Installation</td>
<td>Master Certification Documents List TSA PC12-23.33.02 MCL.</td>
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<td>EASA.A.S.02633</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td>HAWKER HS 125-700 SERIES</td>
<td>HAWKER HS 125-600 SERIES</td>
<td>HAWKER 800 BAE 125-800 SERIES</td>
<td>Installation of a Universal UNS-1F and / or UNS-1L Flight Management System for Basic RNAV, Precision RNAV Operations and Non-precision Approaches and also as a primary means of navigation for oceanic/remote operations.</td>
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<td>EASA.A.S.02634</td>
<td>JET AVIONICS AVIATION &amp; MARKETING SERVICE CENTER s.r.l.</td>
<td></td>
<td>BOMBARDIER CL-215-6B11</td>
<td>Installation of Radar Altimeter AA-300 System</td>
<td>Master Drawing List BDI 002-06-01AFMS BDI 002-06-14</td>
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<td>EASA.A.S.02635</td>
<td>AIRBUS ATLANTIC</td>
<td>REV 1</td>
<td>AIRBUS A320-212 AIRBUS A320-211</td>
<td></td>
<td>Emergency locator transmitter Thales/ Serre Kernard 40K ATP installation Note: This revision reflects the transfer of this STC from EADS Sogerma Services to Sogerma Services.</td>
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<td>EASA.A.S.02636</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV. 4</td>
<td>AIRBUS A318-112</td>
<td></td>
<td>Structural Floor Panel Reconfiguration/ The limitation for divans that they must not be occupied for taxi, take off and landing has been suspended for 1315 FH or 14 months whatever occurs first, starting at June 20th, 2008. The modification is described in the Alert Engineering Bulletin A318- EB03-0000 R...</td>
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<td>Model</td>
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<td>Limitations/Conditions</td>
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<td>EASA.A.S.02839</td>
<td>DASSAULT FALCON SERVICE</td>
<td>MYSTERE FALCON 900</td>
<td>MODE S TRANSPONDER WITH ENHANCED SURVEILLANCE CAPACITY</td>
<td>Master Document List DFS-DE-17-F9-5177 Issue B1 or later approved revisions/Flight Manual Supplement DFS NF 25, Rev Original or later approved revisions</td>
<td>This STC is approved only for those Mystere Falcon 900B aircraft equipped with two Collins TDR-840(-007) Mode &quot;S&quot; ATC Transponders and &quot;Flight Ident&quot; source. Compatibility with previous approved modifications shall be determined by the installer. JAA Leaflet TGL 13 Certification of Mode S Transponder...</td>
<td>30/11/2006</td>
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<td>EASA.A.S.02840</td>
<td>EAD AEROSPACE</td>
<td>B767-300</td>
<td>Installation of Thales Integrated Standby Flight Display (ISFD)</td>
<td>EAD Aerospace 01108-A Flight Manual Supplement 01108-EAS-A-ATM-01 Rev A or later EASA Approved Revision.</td>
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<td>05/03/2007</td>
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<td>EASA.A.S.02841</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING WALTER STARZACHER Gmbh</td>
<td>550</td>
<td>Supplemental Type Certificate for Cargo Net Installation on C550 MSN 0001-0626</td>
<td>EO M04600101</td>
<td>CRI A-02; Registration: OE-GAD; S/N C550-0071</td>
<td>12/02/2007</td>
<td>Active</td>
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<td>EASA.A.S.02842</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING WALTER STARZACHER Gmbh</td>
<td>551</td>
<td>Supplemental Type Certificate for Cargo Net Installation on C551 MSN 0001-0549</td>
<td>EO M04600101</td>
<td>CRI A-02; Registration (if applicable): OE-FAW; Serial number (if applicable): C551-0496</td>
<td>13/04/2007</td>
<td>Active</td>
<td>10014681</td>
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<td>EASA.A.S.02843</td>
<td>DASSAULT FALCON SERVICE</td>
<td>FALCON 10</td>
<td>Extended Function (P-RNAV / RNP 1 Compliances) on Flight Management System (FMS) UNS-1K+ installation</td>
<td>• DFS Master Document List DFS-DE-17-F1-1279 Issue A1; Aircraft Flight Manual Supplement DFS F10-027 Revision Original or later approved revisions for those Falcon 10 aircraft on which this STC has been embodied or later approved revisions of these documents.</td>
<td>his STC is applicable to Falcon 10 aircraft equipped with: • Collins Autopilot and Flight Director AP80; • Collins EFIS 85C-14; • 1 FMS Universal UNS-1K+ with software version 852.2 or later/FMS Universal UNS-1K+ installation has to be embodied through Dassault Falcon Service’s STC DFS 1264 (a...</td>
<td>29/07/2008</td>
<td>Active</td>
<td>10014682</td>
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<td>EASA.A.S.02847</td>
<td>CARGOLUX AIRLINES INTERNATIONALS A.</td>
<td>BOEING 747-400 SERIES</td>
<td>Installation of an Artex B406-A Automatic Fixed Emergency Locator Transmitter on B747-400/600 in accordance with Cargolux Service Bulletin CLX-SB-747-23-001 and Master Drawing list CLX-STC-06-01.</td>
<td>The publications affected as listed in the Service Bulletin/ S6514/001 are to be updated using the information provided in report CLX-STC-06-01.</td>
<td>This STC is approved only for those Mystere Falcon 50 aircraft equipped with: • Two Collins TDR-840(-007) Mode &quot;S&quot; ATC Transponders and &quot;Flight Ident&quot; source, and • Two Inertial Reference Systems. Compatibility with previous approved modifications shall be determined by the installer. JAA Leaflet TGL 13 Certification of Mode S Transponder...</td>
<td>14/03/2007</td>
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<td>EASA.A.S.02848</td>
<td>DASSAULT FALCON SERVICE</td>
<td>MYSTERE FALCON 50</td>
<td>MODE S TRANSPONDER WITH ENHANCED SURVEILLANCE CAPACITY</td>
<td>S62 Master Document List DFS-DE-17-5026 Issue A1 or later approved revisions/Aircraft Flight Manual Supplement DFS NF 32, Rev Original or later approved revisions</td>
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<td>29/11/2006</td>
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<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>DASSAULT FALCON SERVICE</td>
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<td>MYSTERE FALCON 20-FS</td>
<td>Mode S Transponder with enhanced surveillance capacity.</td>
<td></td>
<td>Master Document List DFS-DE-17-F3-839 Issue A1 or later approved revisions. Aircraft Flight Manual</td>
<td>1. This STC is approved only for those Fan Jet Falcon and Mystere Falcon 20 (FS) aircraft equipped with two Collins TDR-943(-007) Mode “B” ATC Transponders and “Flight Ident” source. 2. JAA Leaflet TGL 13 “Certification of Mode S Transponder Systems for Elementary Surveillance” and JAA NPA 20-12a...</td>
<td>18/01/2007</td>
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<td>EASA.A.S.02851</td>
<td>DASSAULT FALCON SERVICE</td>
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<td>MYSTERE FALCON 900</td>
<td>CABIN REFURBISHMENT</td>
<td></td>
<td>Master Document List DFS-DE-17-F9-052 Issue A1 or later approved revisions</td>
<td>This STC is approved only for Mystere Falcon 900 Serial Number Aircraft. 054. Compatibility with previous approved modifications shall be determined by the installer.</td>
<td>21/12/2006</td>
<td>Active</td>
<td>1014688</td>
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<td>EASA.A.S.02864</td>
<td>BOURNEMOUTH AEROSPACE (CONSULTANTS) Ltd.</td>
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<td></td>
<td>BOEING 757-200</td>
<td>Installation of B/C LOPA with standing bar tables</td>
<td></td>
<td>Change Number J438CMC J438PR3016</td>
<td>08/03/2007</td>
<td>Active</td>
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<td>EASA.A.S.02865</td>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td></td>
<td>A340 ALL MODELS</td>
<td>Installation of Aeromobile (ARINC/Telenor Joint Venture) GSM Connectivity System.</td>
<td>1. Master Data List 0300-01-A-2103-R00.2 Certification Plan 0300-01-A-2101-R00.3 System Technical Specification 0300-01-A-2003-R02.4 Compliance Record 0300-01-A-2104-R01.5 Aircraft Flight Manual Supplement 0300-01-A-2301-R00.6 Instructions for continued A/W 0300-01-A-2211-R01...</td>
<td>1. The applicability of this STC is limited to the Airbus A340-313 and A340-541 and any other A340 shown to the identical A340-313 MSN 139 or A340-541 MSN 457, through the DOA's process as defined within the Document 0001-01-B-0614 (Classification of Modification Change Procedure for GSM Connecti...</td>
<td>05/02/2008</td>
<td>Active</td>
<td>10014698</td>
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<td>EASA.A.S.02870</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CO RPRATION AG (AIEC)</td>
<td>REV. 2</td>
<td></td>
<td>BEECH B200 SUPER KING AIR</td>
<td>Installation of a UASIC Flight Management System (FMS) UNS-1L</td>
<td>Installation to be in accordance with AEC Engineering Order ref CV831-862, rev. B dated 19/01/2007 or later approved revision Operation to be in accordance with AEC Aircraft Flight Manual Supplement ref CV831-871, rev. I dated 22/01/2007 or later approved revision Continued Airworthiness to be in...</td>
<td>1. This STC approves only the installation of the parts contained in the above section “Description of Type Design Change”. This STC grants no operational approval to this installation, that will need to be obtained from the State of registry for the particular aircraft involved as defined in JAR OP...</td>
<td>24/01/2007</td>
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<td>EASA.A.S.02872</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 2</td>
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<td>HAWKER HS 125-700 SERIES HAWKER 800XP HAWKER 800</td>
<td>Mode S Transponder Upgrade for existing Collins transponder system.</td>
<td>Upgrade of existing Collins transponder system in accordance with Document List 2036079 DOC and Engineering Order 2036079-EO. Operation in accordance with Flight Manual Suppl. 2036079-AM-FLIGHT Manual Suppl. 2036079-AM-BContinuing Airworthiness 2036079-CA...</td>
<td>1. This STC approves only the installation of the parts contained in the above section “Description of Type Design Change”. This STC grants no operational approval to this installation, that will need to be obtained from the State of registry for the particular aircraft involved as defined in JAR OP...</td>
<td>06/03/2007</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>FAIRCHILD SWEARINGEN SA 227-BC FAIRCHILD SWEARINGEN SA 227-AC FAIRCHILD SWEARINGEN SA 227 FAIRCHILD SWEARINGEN SA 226-TC</td>
<td>Installation of a Garmin GPS 400 System with Remote Annunciator Switch Panel in Fairchild Swearingen SA226-TC, SA227-AC, SA227 BC</td>
<td>Technical Order, Doc. No. 2036263-TO, issue 3 or later approved issueEASA approved Aircraft Flight Manual Supplement, Doc. No. 2036263-AM, issue 2, dated December 2006, or later approved issueInstructions for Continued Airworthiness, Doc. No. 2036263-CA, Issue 1, dated 16- Oct-2006, or later released...</td>
<td>This change is limited to following configuration of the aircraft / equipment type: - Avionics system must be either ARINC429, RS232 or Grey code as defined in the GPS 400 installation manual, and - any model of Rockwell Collins VIR-30/31/32 VOR/ILS</td>
<td>14/12/2006</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>AVIATION STRAUBING ENTWICKLUNGS GMBH</td>
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<td>B.65-90, 65-A90, B90, C90, E90 A, C90A</td>
<td>Dual Installation of COMNAV/GS/GPS CNS430, with DDC31 Roll Manning Converter; Installation of Stormscope WX500 and TCAS Aviwide TAS610.</td>
<td></td>
<td></td>
<td>Prior to installation of this modification the installer must determine that the</td>
<td>12/07/2007</td>
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<td>AERODIENST OMBR &amp; CO. KG</td>
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<td>31</td>
<td>FMS UNS-1L System Installation (UASC)</td>
<td></td>
<td></td>
<td>Certification basis of the change FAS 25, additionally JA Leaflet No. 10, Precision</td>
<td>21/03/2007</td>
<td>Active</td>
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<td>TENENCIA Ltd.</td>
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<td></td>
<td>BAE 146/RJ SERIES/BAE 146/AVRO 146 SERIES</td>
<td>Installation of Cockpit Door Locking System. The modification to the Cockpit Door provides the ability for the Cabin Crew to access the Cockpit in the case of Pilot Incapacitation.</td>
<td></td>
<td></td>
<td>BAE Systems Modification HCM06303C</td>
<td>12/12/2006</td>
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<td>FR AVIATION LTD/DIBA COBHAM AVIATION SERVICES UK</td>
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<td>DORNIER 228-202K</td>
<td>Installation of Rockwell Collins TCAS 4000 system including Dual Mode XSPDR.</td>
<td></td>
<td>- Installation must be in accordance with Master Document List and the Design Change Statement DC-0100, Iss. I, dated 28th March 2007, or later and approved revisions. - Operation must be in accordance with Pilots Operating Handbook Supplement No. FRAHFM(R) 113, Rev. 0, dated 28th March 2007 or later.</td>
<td>1. Compatibility of this modification with other previously approved modifications must be determined by the installer 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation&quot;.</td>
<td>30/03/2007</td>
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<td>LUFTHANSA TECHNIK AG</td>
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<td>AIRBUS A300 MODEL B4-603</td>
<td>A300B4-603 Cabin Reconfiguration from current versions to Intercontinental Layout: Cabin Reconfiguration from current versions (total 202 and 222 Passengers) to Intercontinental Layout (total 217 Passengers).</td>
<td></td>
<td>Compliance Record List CRL25-65Q4M, Issue 1 dated 22 January 2007</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation&quot; shall be determined by the installer.</td>
<td>25/01/2007</td>
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<td>DASSAULT FALCON SERVICE</td>
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<td>FAN JET FALCON SERIES G</td>
<td>Incorporation of Runway Awareness and Advisory System (RAAS) Function on EGPWS Honeycomb Mark V or Mark VII.</td>
<td></td>
<td>Compliance Checklist DFS2576-CC Issue A1 or later approved revisions: Master Document List DFS-DE-17, F2-2576 Issue A1 or later approved revisions; Airplane Flight Manual Supplement DFS N°53 Revision original or later approved revisions for those aircraft previously incorporating EGPWS equipment not b...</td>
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<td>18/07/2007</td>
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<td>EASA.A.S.02893</td>
<td>AIRTOURING LTD</td>
<td>EASA.A.S.02893</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>Installation of Honeywell KGPS60 Class B TAWS</td>
<td>Air Touring Modification AT_MOD_1104</td>
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<td>HB-FPI S/N 551. Valid for this S/N only.</td>
<td>12/12/2006</td>
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<td>EASA.A.S.02895</td>
<td>CZECH AIRLINES TECHNICS A.S.</td>
<td>EASA.A.S.02895</td>
<td>AIRBUS A320-214</td>
<td>Cabin layout modification from 156 passengers and doghouses, installation of new seats for 162 passengers, new doghouses, relocation and addition of PSU's as required, installation of additional attachments for Movable Class.</td>
<td>Modification No. M/20 – 26 /0002</td>
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<td>19/03/2007</td>
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<td>EASA.A.S.02896</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>EASA.A.S.02896</td>
<td>BOEING 757-200</td>
<td>Installation of ATC Mode S Elementary and Enhanced Surveillance</td>
<td>Modification Definition File, MF-B757-06-001, Rev. 00, Date25/03/2007/AFM Suplement, AFMS-8757-06-001, Rev. 00, 15/01/2007/Instructions for Continued Airworthiness, ICA-8757-08-001, Rev. 00, 03/01/2007</td>
<td>Certification basis as FAA TCDS 21N/Acluding:TGL 13 Rev 1,01/06/2003/EASA AMC-30-13. 02/12/2005 JAA TGM 21/07, 01/10/2003.</td>
<td>28/03/2007</td>
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<td>EASA.A.S.02897</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>EASA.A.S.02897</td>
<td>A330-300</td>
<td>Removal of Antenna and Receiver Transmit System and Network Services</td>
<td>Supplemental Type Definition Document, Ref. STD000-0616M, Issue 1, Rev. 01 dated 31 May 2006/Instructions for Continued Airworthiness, Ref. ICA-R-00-07/062, Rev. IR dated 10 April 2007</td>
<td>The AFM Supplement Doc. No. A-00-04/5866 can be removed with the installation of this STC.</td>
<td>31/05/2007</td>
<td>Active</td>
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<td>EASA.A.S.02898</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>EASA.A.S.02898</td>
<td>AIRBUS A340-300</td>
<td>Removal of Antenna and Receiver Transmit System and Network Services</td>
<td>Supplemental Type Definition Document, Ref. STD000-0617M, Issue 01, Rev. 00 dated 1 June 2007/Instructions for Continued Airworthiness, Ref. ICA-R-00-07/064, Rev. IR dated 10 April 2007</td>
<td>The AFM Supplement Doc. No. A-00-04/4773 can be removed within the-installation of this STC.</td>
<td>19/06/2007</td>
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<td>EASA.A.S.02899</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>EASA.A.S.02899</td>
<td>AIRBUS A340-600</td>
<td>Removal of Antenna and Receiver Transmit System and Network Services.</td>
<td>Supplemental Type Definition Document, Ref. STD000-0617M, Issue 01, Rev. 00 dated 1 June 2007/Instructions for Continued Airworthiness, Ref. ICA-R-00-07/065, Rev. IR dated 10 April 2007</td>
<td>The AFM Supplement Doc. No. A-00-04/4773 can be removed within the-installation of this STC.</td>
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<td>EASA.A.S.02900</td>
<td>CAA UK RESPONSIBILITY</td>
<td>FOKKER F.27 MARK 700, FOKKER F.27 MARK 600, FOKKER F.27 MARK 500, FOKKER F.27 MARK 400, FOKKER F.27 MARK 300, FOKKER F.27 MARK 200, FOKKER F.27 MARK 100, FOKKER F.27 MARK 0604, FOKKER F.27 MARK 0502, FOKKER F.27 MARK 050</td>
<td>Introduction of Rockwell Collins DualADF-40 System in the F27 Mark 200. This installation replaces the existing BendixDual ADF System in accordance with STC Twenty One Limited Master Drawing List document S21. MDL-0264 dated 14/12/2006 or later EASA approved revision.</td>
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<td>EASA.A.S.02901</td>
<td>AVIATION DEFENSE SERVICE</td>
<td>FalCon 10</td>
<td>AVD-MDA10-34-060A: installation of EGPWS Mark VIII</td>
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<td>EASA.A.S.02902</td>
<td>FRANCE AVIATION</td>
<td>BEECH F33</td>
<td>Installation and BRNAV approval of Garmin GNS-480.</td>
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<td>EASA.A.S.02903</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GMBH</td>
<td>CESSNA C650</td>
<td>Installation of Solid State Flight Data Recorder (SSFDR) F1000</td>
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<td>EASA.A.S.02906</td>
<td>JET AVIATION VIENNA GMBH</td>
<td>S5C</td>
<td>Upgrade of the SSR Transponder System to Elementary (Enhanced) Surveillance</td>
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<td>EASA.A.S.02907</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>CESSNA 560 (CITATION V), CESSNA 550 CITATION II, CESSNA 550 BRAVO, CESSNA 550, CITATION</td>
<td>Mode-S upgrade for existing Honeywell MST 67A transponder system.</td>
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**Certification basis**: FAR 25 Amdt 20. **Limitations/Conditions**: IAW AFM Supplement Supavd1 to DTM 721 apply for operation with EGPWS installation IAW AVD-MDA10-34-060A. TGL12 complied with. Cert basis FAR 25 Am11-20.

**Date of Issue**: 20/06/2007. **Status**: Active. **System Record**: 10014722.
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<td>EASA.A.S.02909</td>
<td>AVIATION DEFENSE SERVICE</td>
<td>AVD-MDA20-95-057A: Activation of the SETHI 2 mission</td>
<td>see Change Approval Sheet CAS AVD-MDA20-95-057A Ed 1 dated 21.6.07 or later approved revisions, see Change Approval Sheet CAS AVD-MDA20-95-057A issue 1 dated 21.11.08 and associated technical documentation. AFM supplement SAFM,FALC20 rev 15 dated 17.05.08 or later approved revisions, and associated technical documentation.</td>
<td>18/12/2008</td>
<td>Active</td>
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<td>MYSTERE-FALCON 20</td>
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<td>1. This STC ensures proper isolation of Payload (SETHI Mission Equipment) from return to civil configuration in case of inadvertent activation. 2. Activated SETHI Mission Equipment is not approved by EASA. 3. Limitations see AFM Supplement SAFM,FALC20 to DTM 919 apply for operation with SETHI equi...</td>
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<td>EASA.A.S.02913</td>
<td>COMPAGNIA GENERALE RIPRESEARDEES.p.A. - CGR S.p.A.</td>
<td>REIMS AVIATION CESSNA F-172 CESSNA 172P</td>
<td>CERI P001-00 &quot;Structure &amp; Exhaust System Modification for Camera Installation&quot;.</td>
<td>AFM Supplement CERI P001-00 ICA Applicable to S/N 172-7631 only.</td>
<td>04/06/2008</td>
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<td>EASA.A.S.02914</td>
<td>PHOENIX AEROSPACE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td>1. Restricted to G-PPLC ASN 560-0059 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft/ engine configurations shall be determined...</td>
<td>24/01/2007</td>
<td>Active</td>
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<td>EASA.A.S.02915</td>
<td>DASSAULT FALCON SERVICE</td>
<td>MYSTERE FALCON 50</td>
<td>Mode S Transponder with enhanced surveillance capacity.</td>
<td>Master Document List DFS-06-17-FS-0630 Issue A1 or later approved revisions Aircraft Flight Manual Supplement DFS NF 56 Rev Original or later approved revisions. This STC is approved only for those Mystere Falcon 50 aircraft equipped with two Collins TDR-94D(-007) Mode &quot;S&quot; ATC Transponders, &quot;Flight Idle&quot; source and without IRS and with one or two FMS’s Compatibility with previous approved modifications shall be determined by the installer JAA Leaflet TGL 13...</td>
<td>08/02/2007</td>
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<td>EASA.A.S.02918</td>
<td>AERODISA S.L.</td>
<td>MYSTERE FALCON 20</td>
<td>AED2006-14M: TAWS installation for Falcon 20.</td>
<td>Master Data List AED2006-14M MDL Ed 06, Rev. 3 dated 22-07-2007 or later EASA approved revisions, FM Supplement AED2006-14M_D03 Ed 06, Rev. 2 dated 12-07-2007 or later EASA approved revisions. See FM Supplement AED2006-14M_D03 Section II &quot;Limitations&quot; Certification basis: CS25 initial issue, TGL 12 complied with SANDEL mandatory SB 3400-09 to be incorporated.</td>
<td>26/07/2007</td>
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<td>EASA.A.S.02919</td>
<td>E.I.S.A.</td>
<td>CEESNA 525A</td>
<td>Installation of SAT-COM Antenna</td>
<td>EISA 01 SLO 5120, EISA 01 SSTD S120</td>
<td>24/01/2007</td>
<td>Orphaned</td>
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<td>EASA.A.S.02921</td>
<td>JET AVIATION AG, BASEL</td>
<td>MYSTERE FALCON 50</td>
<td>MK-V EGPRS upgrade.</td>
<td>• Master Data List Doc No: BFA5004462-001 • AFMS Doc No: BFA53034462-201-01. 1. Required Avionica Configuration: Collins ProLine 98 Collins EFIS 985958-0 Colpix WIP 8502. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/ engine configuration.</td>
<td>04/01/2007</td>
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<td>EASA.A.S.02922</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING RIZATION TECH. INC</td>
<td>BEECH B300</td>
<td>Standby Instruments Installation: 2&quot; Revere Thrommer Standby Altimeter &amp; 2&quot; Revere Thrommer Aspeed Indicator &amp; L3 Communications Standby Altitude Indicator AI-804 with Goodrich JET Emergency Power Supply Battery Pack PF-835</td>
<td>CV 596-049</td>
<td>11/05/2007</td>
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<td>AIRTOURING LTD</td>
<td>CAA UK RESPONSIBILITY</td>
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<td>B200</td>
<td>Installation of KGP560 TAWS and Dual GTX330D Mode S</td>
<td>At Touring Mod Sheet AT.MOD.1105Compliance Record AT_CMRO_1105Flight Manual Supplements AT. FMS.1105 Rev A &amp; AT. FMS.1105-1 Rev A or later EASA approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.A.S.02924</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
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<td>BEECH 1900D BEECH 1900C WET WING, Airliner BEECH 1900</td>
<td>Installation of an Executive Power System (EPS) with Inverter and Power Outlet.</td>
<td>• Installation Procedure: 2006-10-1 Rev. B • AFM-Supplement: 2006-10-8 Rev.C.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.A.S.02925</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
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<td>BEECH 1900D BEECH 1900C, Airliner BEECH 1900</td>
<td>Installation of a Satcom System for Passenger Communication.</td>
<td>• Installation Procedure: 2006-9-1 Rev. B • AFM-Supplement: 2006-9-8 Rev. A</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES.A.R.L.</td>
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<td>BEECH V35B</td>
<td>Modification d'installation BRNAV.</td>
<td>FAM 1348/06/SEAM/AFM Supplement n° 1, revision 0, dated 8 December 2006</td>
<td>AFM Supplement n°1, revision 0, dated 8 December 2006 or any later approved revision must be on board.</td>
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<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td>BAE ATP</td>
<td>Lead Acid Battery Installation.</td>
<td>- TEN 494 Issue 4 - Report TEN 494/D Issue 2 - Report TEN 494/1 Compliance Report Issue 1</td>
<td>1. Applicable only to aeroplanes that are fitted with anappropriately approved Class ECargo Compartment used only for the carriage of cargo.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously...</td>
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<td>ELECTRONIQUE MAINTENANCE AVIATION</td>
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<td>FARICHLD SWERINGEN SA 226</td>
<td>Garmin GNS 530A VHF 8.33KHZ, Honeywell KY 196B VHF 8.33 KHZ, BF Goodrich TAS SKY 699 Installation</td>
<td>Airplane Flight Manual Supplement ref. TS21-899 revision IR, dated August 19, 2003 or later approved revision.</td>
<td>This STC is limited to the SA226 Sn T312</td>
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<td>EASA.A.S.02936</td>
<td>TRANSARCO S.A.</td>
<td>MYSTERE FALCON 900</td>
<td>STC Holder</td>
<td>MODE S TRANSPONDER WITH ENHANCED SURVEILLANCE CAPACITY</td>
<td>Master Certification Document List TSA F900-34.80.00, MCDL Rev. Original or later approved revisions. Airplane Flight Manual Supplement TSA F900-34.80.03, AFM Supplement No. BB72734461-001. Maintenance Manual Supplement TSA F900-34.80.03, MMS, Rev. Original or later approved revisions.</td>
<td>This STC is approved only for those Mystere Falcon MF-900 aircraft excluding Falcon 900C version and incorporating at least: • Dual Collins TDR-94D Mode “S” ATC Transponders compliant with Elementary Surveillance and/or Single or dual GPS receiver Compatibility with previous approved modification...</td>
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<td>AEROCONDOR TRANSPORTES AEREOS SGPS</td>
<td>SD3-60</td>
<td>STC Holder</td>
<td>Installation of TAWS</td>
<td>Certification Programme STC ATA0601. Issue: DOAFM - ATA.021J.081.0601.01</td>
<td>As per AFM</td>
<td>EASA.A.S.02940</td>
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<td>EASA.A.S.02942</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>BAE.125 SERIES 800B</td>
<td>STC Holder</td>
<td>TAWS ST3400 System Installation (Sandel)</td>
<td>Installation in accordance with Aero-Dienst Engineering Order ref 229-HV, Rev. IR dated 17. Jan. 2007, or later approved revision. To be operated in accordance with Aero-Dienst Airplane Flight Manual Supplement ref AG636-071, Rev IR or later approved revision.</td>
<td>Operations must be in accordance with Flight Manual Supplement Aero-Dienst 7606 SAFM Rev. 01, dated 17-08-2007, or later approved revisions. Maintenance must be in accordance...</td>
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<td>EASA.A.S.02944</td>
<td>AEROTEC &amp; CONCEPT</td>
<td>208, 208B</td>
<td>STC Holder</td>
<td>Video Mast Installation under Cessna 208.</td>
<td>Installation must be in accordance with Aerotec Master Document List 7606 MDL Rev. 02, dated 22-08-2007, or later approved revisions. Operations must be in accordance with Flight Manual Supplement Aerotec 7606 SAFM Rev. 01, dated 17-08-2007, or later approved revisions. Maintenance must be in accordance...</td>
<td>EASA.A.S.02944</td>
<td>Installation must be in accordance with Aerotec Master Document List 7606 MDL Rev. 02, dated 22-08-2007, or later approved revisions. Operations must be in accordance with Flight Manual Supplement Aerotec 7606 SAFM Rev. 01, dated 17-08-2007, or later approved revisions. Maintenance must be in accordance...</td>
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<td>TENENCIA Ltd.</td>
<td>CAIA UK RESPONSIBILITY</td>
<td>AIRBUS A300 MODEL B2 AIRBUS A300 B4</td>
<td>Certification basis</td>
<td>Description</td>
<td>Limited/Conditions</td>
<td>Date of Issue</td>
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<td>TEN/A300/001</td>
<td>TEN/A300/001</td>
<td>TEN/A300/001</td>
<td>TEN/495/Rev A</td>
<td>Installation of dual Universal UNI-1L or UNI-1k Flight Management System (FMS) / This modification allows complying with Precision RNAV operations in designated European airspace.</td>
<td>The installation of this STC has a prerequisite aircraft configuration. This STC is only applicable on aircraft 1, in which the following modifications have been embodied: FAA STC ST07113CH, accepted by EASA under the grandfather rights ruling of the CAA UK.</td>
<td>22/02/2008</td>
</tr>
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EASA STC no. | SCANDINAVIAN AVIONICS DESIGN A/S | TENENCIA Ltd. | AIRBUS A300 MODEL B2 AIRBUS A300 B4 | Certification basis | Description | Limited/Conditions | Date of Issue | Status | System Record |
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<td>TEN/A300/001</td>
<td>TEN/495/Rev A</td>
<td>Replacement of existing Transponder System by a Garmin GTX 330D Mode-S Transponder System</td>
<td>As defined in Scandinavian Avionics Modification ref 2036080</td>
<td>12/02/2007</td>
<td>Active</td>
<td>10014759</td>
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EASA STC no. | AVIATION DEFENSE SERVICE | TENENCIA Ltd. | AIRBUS A300 MODEL B2 AIRBUS A300 B4 | Certification basis | Description | Limited/Conditions | Date of Issue | Status | System Record |
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<td>TEN/A300/001</td>
<td>TEN/495/Rev A</td>
<td>Installation of two 8.33 kHz VHF</td>
<td>None</td>
<td>This STC is limited to the S/N BB199 and BB202</td>
<td>04/04/2007</td>
<td>Active</td>
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EASA STC no. | CESSNA 421C | CESSNA 421C | CESSNA 421C-200 | Certification basis | Description | Limited/Conditions | Date of Issue | Status | System Record |
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<td>TEN/A300/001</td>
<td>TEN/495/Rev A</td>
<td>Upgrade of existing M67A</td>
<td>None</td>
<td>This STC is only approved only for the product configuration as defined in the approved design data referred to in paragraph 'Description' Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>02/04/2007</td>
<td>Active</td>
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EASA STC no. | CESSNA 425 | CESSNA 425 | CESSNA 425 | Certification basis | Description | Limited/Conditions | Date of Issue | Status | System Record |
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<td>TEN/A300/001</td>
<td>TEN/495/Rev A</td>
<td>Replacement of existing Transponder System by a Garmin GTX 330D Mode-S Transponder System</td>
<td>None</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in paragraph 'Description' Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>11/05/2007</td>
<td>Active</td>
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EASA STC no. | TENENCIA Ltd. | CAIA UK RESPONSIBILITY | BOEING 737 | Certification basis | Description | Limited/Conditions | Date of Issue | Status | System Record |
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<td>TEN/A300/001</td>
<td>TEN/A300/001</td>
<td>TEN/495/Rev A</td>
<td>Installation of Mode S Transponders for Enhanced Surveillance.</td>
<td>None</td>
<td>This STC is limited to installation on Boeing 737-852 S/N 29029 only.</td>
<td>31/05/2007</td>
<td>Active</td>
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<td>STC Holder</td>
<td>CAA UK RESPONSIBILITY</td>
<td>Model</td>
<td>Description</td>
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<tr>
<td>EASA.A.S.02966</td>
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<td>FAN JET FALCON</td>
<td>Non Capable Enhanced Mode S CEHS Transponder Installation Limitation/hardwires/limit/ \ capacity restricted to a subset of all required enhanced mode S parameters</td>
<td>EASA.A.S.02960, 02961, 02962</td>
<td>01/06/2007</td>
<td>03/07/2007</td>
<td>Active</td>
<td>1014319</td>
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| EASA.A.S.02967              |                       | DORNIER 228-202, SIN     | Enhanced ATC Mode S: Dornier Qualification of existing Elementary Mode S & ATC systems to Enhanced Mode S
requirements with limited DAP set.                                                                                      | EASA.A.S.02967, 02968, 02969, 02970                                                    | 01/06/2007                                                                                   | 03/07/2007     | Active | 1024320       |
| EASA.A.S.02959              |                       | CESSNA 550, CESSNA 550 C1 | Installation of Honeywell Multi-Function Radar (MFRD) and an optional Honeywell Primus Radar System, if the existing radar is incompatible with the MFRD. | Compliance Report 2038716-3, Issue 2AFM Supplement 2038716-AM, Issue 2 Instructions for Continued Airworthiness
2038716-CA, Issue 1 for later approved versions.                                                                 | 2038716-3, Issue 2AFM Supplement 2038716-AM, Issue 2 Instructions for Continued Airworthiness
2038716-CA, Issue 1 for later approved versions.                                                                 | 16/02/2009     | Active | 1014176       |
| EASA.A.S.02960              |                       | CESSNA 425              | Installation of Traffic Advisory System                                                                                       | Leas Avionics Modification HL/MOD/9686TC - Definition Document OD 799001 Issue 2AFM -
AFMS 799001 Issue 1 for later approved versions.                                                                 | As defined in the AFM supplement.                                                                 | 08/03/2007     | Active | 1014176       |
| EASA.A.S.02961              |                       | CESSNA 425              | Installation of Garmin GNS 530 & GTX4330                                                                                     | Leas Avionics Modification HL/MOD/9686TC - Definition Document OD 799001 Issue 2AFM - AFMS 799001 Issue 1 for later approved versions. | As per AFM Supplements.                                                                 | 08/03/2007     | Active | 1014176       |
| EASA.A.S.02962              |                       | 747-400                 | Interior Changes required on Cathay Pacific Boeing 747-400 for BE Aerospace Flight Structures.                              | BBJ Design Services Modification reference BBJ866.                                      | Prior to installation of this modification the installer must determine that the
interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third parties... | 02/07/2007     | Active | 1014176       |
20-13, 22/12/2006                                                                 | 03/04/2007     | Active | 1014177       |
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<th>Status</th>
<th>System Record</th>
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<td>EASA.A.S.02964</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>STC Holder</td>
<td>SA26-AT, SA26-T, SA227-AT, SA227-RT, SA226-AT, SA226-T, SA226-T(B)</td>
<td>Installation of Mode S &amp; XFDI Diversity Garmin GTX 330D with now top Antenna</td>
<td>Doc. No. Asr 2008-073-MDL-01-00 Amend. 00 dated 24/06/2001 or later approved</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product</td>
<td>16/10/2007</td>
<td>Active</td>
<td>10014772</td>
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<tr>
<td>EASA.A.S.02965</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>STC Holder</td>
<td>525, 525A</td>
<td>Installation of one Garmin GNS530 in a Cessna 525</td>
<td>Installation must be in accordance with Avionik Straubing Master Document List Doc. No. Asr 2008-021-MDL-01-00 Amend. 00 dated 01.04.08 or later Approved</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/06/2008</td>
<td>Active</td>
<td>10014773</td>
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<td>EASA.A.S.02966</td>
<td>BBJ DESIGN SERVICES LIMITED</td>
<td>CA4 UK RESPONSIBILITY</td>
<td>767-300</td>
<td>Installation of an in-seat power supply system and reconfiguration of business class seating</td>
<td>BBJ MOD No. BBJ840 Issue 3 or later. Maintenance Manual Supplement BBJ MS057</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>30/05/2007</td>
<td>Active</td>
<td>10014774</td>
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<tr>
<td>EASA.A.S.02973</td>
<td>AIR ALLIANCE GmbH</td>
<td>STC Holder</td>
<td>TBM 700</td>
<td>Installation of Garmin GTX330D Transponder with Top Antenna</td>
<td>Installation and Maintenance Manual 2006_902_0700_500Rev. 07 Flight Manual Supplement 2006_902_0700_510 Rev.2</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft configuration shall be determined by the installer.</td>
<td>15/08/2007</td>
<td>Active</td>
<td>10014775</td>
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<tr>
<td>EASA.A.S.02975</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>STC Holder</td>
<td>CESSNA 650 CITATION VII CESSNA 650 CITATION VI CESSNA 650 CITATION III</td>
<td>Installation in accordance with Scandinavian Avionics Engineering Order Reference 2037690-EO Issue 2, dated 20 April 2007, or later approved revisions.Aircraft to be operated in accordance with Scandinavian Avionics Flight Manual SupplementReference 2037690-AM-A and 2037690-AM-B as required, both at...</td>
<td>Aircraft Serial No. 650-0027 with FOCA STC Z34-44-05 installed as defined in Scandinavian Avionics Modification Reference 2037690.</td>
<td>29/05/2007</td>
<td>Active</td>
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<td>AEROCONDOR TRANSPORTES AEREOS SGPS</td>
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<td>DORNIER 228-202</td>
<td>Installation of Honeywell TAWS Mark VI System with GPS Antenna Honeywell KA-92 and OAT Probe.</td>
<td>Definition Document ATA. 212.81.083.002.01, Issue 1. AFM Supplement No. ATA.212.081.0603.010, Issue 1.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is prohibited.</td>
<td>08/08/2007</td>
<td>Active</td>
<td>10014780</td>
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<td>EASA.A.S.02979</td>
<td>SNS CERTIFICATION SERVICES Ltd.</td>
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<td>MD-11 EMBRAER ERJ 145 SERIES</td>
<td>Installation of STG Aerospace SafeTGlo Photoluminiscent Bulkhead/Divider Exit Signs.</td>
<td>As listed in SWS Certification Ltd. Design report 3001-DR-001, Issue 1 or later EASA approved issue. As defined in SWS Certification Ltd. Design report 3001-DR-001, Issue 1 or later EASA approved issue.</td>
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<td>19/06/2007</td>
<td>Active</td>
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<td>EASA.A.S.02981</td>
<td>AVIATION DEFENSE SERVICE</td>
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<td>MYSTÈRE-FALCON 20 SERIES</td>
<td>Enhanced Surveillance Mode S Transponder installation.</td>
<td>AvDef Modification Sheet N° AVD-MDA-20-60 Rev 5, or later revisions. Falcon 20 AFM Supplement Evaluation N°1 Enhanced Mode S Installation Issue 0 dated 08 June 2007 Or later approved revisions of these documents.</td>
<td>Major change applicable to aircraft Falcon 20 serial numbers 079, 103 and 281 equipped with: - Integrated Collins Proline IV avionics suite - DFS Elementary Mode S Transponder and as per EASA.A.S.00203 STC - AvDef Minor change EASA.A.C.00467 upgrade of IOC 4000 and upgrade of transponder...</td>
<td>16/01/2008</td>
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<td>EASA.A.S.02985</td>
<td>AeroData AG</td>
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<td>B300</td>
<td>Installation of EGPWS Mark VI Upgrade in B300.</td>
<td>Installation in accordance with Master Document Listef MDL-AJC-117 Rev 3 or later revisions Operation in accordance with Flight Manual Supplement EGPWS Mark VI Rev 0 or later revisions Instructions for Continued Airworthiness of the aircraft.</td>
<td>Compatibility of this modification with other previously approved modifications must be determined by the installer.</td>
<td>16/04/2007</td>
<td>Active</td>
<td>10014785</td>
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<td>EASA.A.S.02986</td>
<td>E.I.S.A.</td>
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<td>FALCON 50</td>
<td>ADDITION OF ENHANCED SURVEILLANCE CAPACITY ON EXISTING ELEMENTARY MODE S TRANSPONDER INSTALLATION</td>
<td>Compliance Checklist EISA 06 CCL S119 Revision 2 or later approved revisions.</td>
<td>This STC is approved only for those Mystere Falcon 50 aircraft previously equipped to this modification with: - Option 1: Rockwell Collins Proline IV, FMS FMC-6000 and Mode S Elementary Surveillance Installation with Transponder Rockwell Collins TDR 940 version .008, and - Option 2: FMS UNS-1L, etc...</td>
<td>26/05/2007</td>
<td>Orphaned</td>
<td>10014786</td>
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EASA A.S.02987
S.E.A.M SERVICE ELECTRONIC AVIATION MARINES A.R.L.
Rev. 1
FOCA STC 31-31-05
Installation of new ELS/EHS transponder configuration.

EASA A.S.02988
A Y M AVIONICS, S.L.
Installation of new Honeywell MST-67A Mode S transponder.

EASA A.S.02990
TRANSARIO S.A.
FOCA STC 31-31-05
Upgrade of two Mode-S transponders honeywell MST-67A to honeywell SB S ELS/EHS.

EASA A.S.02991
RUAG AEROSPACE SERVICES GmbH
CESSNA 650 CITATION VII, VI
Upgrade of two Mode-S transponders honeywell MST-67A by honeywell SB to implement ELS/EHS transponder capability. ATC Controller KFS-578A is exchanged by Mode S Controller KFS-578A. New wires from the transponders to the FMS.

EASA A.S.02992
RUAG AEROSPACE SERVICES GmbH
BOMBARDIER CL-600-2B16 (601-3R)
New Supplement to Aircraft Flight Manual & Avionics Document SMV 01/07 R0 Operating & Service Instructions to Implement ELS/EHS transponder Honeywell MST-67A by Honeywell SB to new equipment type as listed under the configuration of aircraft / engine type.

EASA A.S.02993
VITROCSET S.p.A.
REV. 1 EASA A.S. 02993
Installation of TAWS/RMI Sandel ST3440.

EASA A.S.02994
RUAG AEROSPACE SERVICES GmbH
BOMBARDIER CL-600-2A12 (CL-601)
Inflight Entertainment System (IFES) SIRIUS.

EASA A.S.02995
AVIONICS INTEGRATION & ENGINEERING INC.
RPORATION AG (AIEC)
REV 1 BEECH 400A
Upgrade of existing FID capable dual TDR-940 Mode S transponder system to enhanced surveillance capability by incorporation of Collins SB 502 and SB 507.

EASA A.S.02996
AERTOC & CONCEPT
PA-31T (CHEYENNE)
Installation of ATC mode S.

Installation of TAWS/RMI Sandel ST3440
Installation ATC mode S.
Master Data List ref. "STC Upgrade of existing FID capable dual
Associated Technical
AVIONICS
AEROTEC & CONCEPT
PA-31T (CHEYENNE)
Installation of TAWS/RMI Sandel ST3440
Installation ATC mode S.
Master Data List ref. "STC Upgrade of existing FID capable dual
Associated Technical
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<th>Description</th>
<th>Certification basis</th>
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<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td>E100</td>
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<td>BEECH 100</td>
<td>Installation of 2 GNS430A, 2 GTX330D XPDR, 1KMR 675 and 1 GAD42.</td>
<td>Installation must be in accordance with S.E.A.M. MDL 1440-06/MDL/SEAM Ed.1, Rev. 2, dated 29.06.07 or later Approved Revisions. Operation must be in accordance with S.E.A.M. Aircraft Flight Manual Supplement au Manuel de vol for GNS430 SEAM AFM 1440-06-01, Rev. 1, dated 29.06.07 and Supp...</td>
<td>Installation limited to Beech B100, S/N: B-199 (F-GEVX).</td>
<td>20/07/2007</td>
<td>Active</td>
<td>10014792</td>
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<td>EASA.A.S.02998</td>
<td>S.E.A.M.SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td></td>
<td></td>
<td>KING AIR 200</td>
<td>Installation of 2 GNS430A, 2 GTX330D XPDR and 1 GAD42.</td>
<td>Installation must be in accordance with S.E.A.M. MDL 1428-06/MDL/SEAM Ed.1, Rev. 2, dated 29.06.07 or later Approved Revisions. Operation must be in accordance with S.E.A.M. Aircraft Flight Manual Supplement au Manual de vol for GNS430 SEAM AFM 1428-06-01, Rev. 1, dated 29.06.07 and Supp...</td>
<td>Installation limited to Beech B200, S/N: BL-11 (F-GRSO).</td>
<td>10/07/2007</td>
<td>Active</td>
<td>10014793</td>
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<td>EASA.A.S.03002</td>
<td>AEROTEC &amp; CONCEPT</td>
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<td>PIPER PA-31T</td>
<td>Modification radio installation.</td>
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<td>* Master Data List ref. “STC groupe 2 n° 1507”, revision0, dated August 2, 2007 or later approved revision.* Avionics Flight Manual Supplement, ref. 1507-AM, revision 0, dated January 25, 2007 or later approved revision.</td>
<td>Compatibility of this design change with other previously approved modifications must be determined by the installer.</td>
<td>10/09/2007</td>
<td>Active</td>
<td>10014798</td>
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<td>EASA.A.S.03004</td>
<td>LUFTHANSAS TECHNIK AG</td>
<td></td>
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<td>BOEING 737-900</td>
<td>ATC Mode S – Enhanced Surveillance Upgrade</td>
<td>- Supplemental Type Design Definition ST DD34-06/ICM Issue 1, Rev 1, dated 26th February 2007, or later EASA approved revision.- AFM Supplement LHT Doc number A-34-07004-AFM, or later EASA approved revision.- Instructions for Continuous Airworthiness Doc number R-34-07040.</td>
<td>Compatibility of this design change with other previously approved modifications must be determined by the installer.</td>
<td>02/03/2007</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.A.S.03005</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td></td>
<td>525, 525A</td>
<td>Installation of Garmin GTX330D single/dual.</td>
<td>Installation must be in accordance with Master Document List Engineering Order No. AR. 2007-006-00 Rev. 00 dated 25-01-07 or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement No. AR. 2007-006-00. FMS-05-1-00 Rev. 00 dated 25-01-07 or later approved revision.</td>
<td></td>
<td>Limited to Cessna 525, SN 0001-0359. Compatibility of this modification with other previously approved modifications must be determined by the installer.</td>
<td>08/02/2007</td>
<td>Active</td>
<td>10014802</td>
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<td>EASA.A.S.03006</td>
<td>HAWKER BEECHCRAFT Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>HAWKER 850 XP</td>
<td>Introduction of Steep Approach Capability.</td>
<td></td>
<td>Modification SC0281A AFMS Hawker Beechcraft Supplement 1 issue 1 or later approved issue.</td>
<td>03/07/2007</td>
<td>Active</td>
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<td>EASA.A.S.03013</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>BEECH B300C BEECH B300 BEECH 300LW BEECH 300 BEECH 200C BEECH 200</td>
<td>Installation of Universal UMS-1L (-1K+). FMS</td>
<td>Engineering Order, Doc. No. 2037417-EO, Issue 2, dated 09-Feb-2007 or later approved issue/Installation Documentation (Certification Documents), Doc. No. 2037417 STC, Issue 2, dated 12-Feb-2007, or later approved issue/EASA approved Aircraft Flight Manual Supplement, Doc. No. 2037417-AM, Issue 2, dated 09-Feb-2007 or later approved issue.</td>
<td>This change is limited to the configuration of aircraft / equipment type as listed under Item 4. in Engineering Order, Doc. No. 2037417-EO, Issue 2, dated 09-Feb-2007 or later approved issue.</td>
<td>13/02/2007</td>
<td>Active</td>
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<td>EASA.A.S.03015</td>
<td>AVIONITEC AG</td>
<td></td>
<td>501</td>
<td>Replacement of XPSR to GARMIN GTX 330D Diversity Mode S and change of GPS Interface</td>
<td>Master Document List 063269-017, latest approved Revision/Avionitec AFM-S Doc. 063269-012 from 14 November 2007, or later EASA approved Revision/Avionitec AFM-S Doc. 063269-011 from 1 November 2007, or later EASA approved Revision/Avionitec Instruction for continued airworthiness 063269-015 from 06....</td>
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<td>29/01/2008</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.03016</td>
<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>525A</td>
<td></td>
<td>Installation of single/dual Garmin GTX 330D()Mod: Casana 525A, S/N 0001 and on</td>
<td>ASR2007-009-()TCP-09.00, ASR2007-009-()MDL-01-01EASA approved AFM ()S Doc. No. ASR2007-009-()FMS-05.1-00, Issue 1, or later approved RevisionEASA approved AFM ()S Doc. No. ASR2007-009-()FMS-05.2-00, Issue 1, or later approved RevisionEASA approved AFM ()S Doc. No. ASR2007-009-()FMS-05.3-00, Org. Issue, or later revision</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>21/06/2007</td>
<td>Active</td>
<td>10014811</td>
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<td>EASA.A.S.03019</td>
<td>ELECTRONIQUE MAINTENANCE AVIATION</td>
<td></td>
<td>SA226-T, SA-226-AT, SA226-T(B)</td>
<td>Installation of Avionic Upgrade according EMA/07/001/SA226/VHF8.33-ATC-MFD-RDR</td>
<td>Master Doc. List EMA/ 07/001/SA226/VHF8.33-ATC-MFD-RDR</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>30/04/2007</td>
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<td>EASA.A.S.03022</td>
<td>STC TWENTY ONE LIMITED</td>
<td>REV 2</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Introduction of Passenger Cabin interior.</td>
<td>1. STC 21 Master Document List No. 21 MDL-0316 Issue 1 dated 15 January 2008 or later approved revision. 2. STC 21 Flight Manual Supplement No. S21:FMS-0033 Issue 1 or later approved revision.</td>
<td>EASA STC No. EASA IM.A.S. 01752 must be fitted. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The instal...</td>
<td>16/01/2009</td>
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<td>EASA.A.S.03023</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CORPORATION AG (AIEC)</td>
<td></td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Enhanced Surveillance Mode S Transponder Upgrade (Collins)</td>
<td>AEIC Modification ACES2-015 AEIC MDL No. ACES2-049 at Revision A dated March 7, 2007 AEIC AFM Supplement No. ACES2-071 at Revision IR.</td>
<td>N/A</td>
<td>15/03/2007</td>
<td>Active</td>
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<td>EASA.A.S.03025</td>
<td>NORTHERN AVIONICS S.R.L.</td>
<td></td>
<td>PIPER PA-31T</td>
<td>Installation of Mode S Diversity XPDR Honeywell MST-67A (Elementary Surveillance).</td>
<td>Installation must be in accordance with Master Document List included in the Northern Avionics Modification Bulletin NAV-324 Rev 3 dated 1st March 2007 or later Approved Revision. Operation must be in accordance with Northern Avionics Airplane Flight Manual Supplement No. NAV-324 Rev. 0 dated 1st...</td>
<td>13/07/2007</td>
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<td>EASA.A.S.03029</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td></td>
<td>BOMBARDIER CL-600-2B16 (CL-604)</td>
<td>EASA Recertification of Cabin Modifications.</td>
<td>RUAG Master Document List No. 1450-MDL</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>18/10/2007</td>
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<tr>
<td>DASSAULT FALCON SERVICE</td>
<td>MYSTERE-FALCON 200</td>
<td>DFS-DE-17-F4-H28 Issue 3 Mylare Falcon 200 AFM Supplement DFS 12 Original Issue Maintenance Manual Supplement + Supplement S-2034 Revision 1 - SMIM4-95-0642 and SMIM 34-95-1542 Revision 1</td>
<td>This STC is limited to all Mylare Falcon 200 aircraft equipped with two Mode S Transponders Honeywell MIST-67A with antenna diversity as part of the original transponder installation plus two FMS Universal (UNS-1) and two ADC 80 installed. See Mylare Falcon 200 AFM Supplement DFS 12 Original Issue Maintenance Manual Supplement + Supplement S-2034 Revision 1 - SMIM4-95-0642 and SMIM 34-95-1542 Revision 1</td>
<td>22/04/2008</td>
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<td>1015623</td>
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<td>COMPAGNIE FRANCE AVIATION</td>
<td>R2160</td>
<td>AFM-07-2007 Edition 1 - Installation Instruction sur R2160</td>
<td>none</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>28/07/2009</td>
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<td>1015272</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>BOEING 767-300</td>
<td>Air France Change Reference CAS-767-04-001 Rev 1.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>26/02/2007</td>
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<td>1014422</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>CESSNA 500, CITATION</td>
<td>STC Document List 2038635 STC Compliance Record AMC 2038635- CRAFM Supplement ELS 2038635-AM-AAFM Supplement ELS 2038635-AM-B</td>
<td>According to EUROCONTROL definition this equipment with approval EASA A.S.03036 is non-EHS compliant. Prior to installation of this modification the installer must determine that the internaship between this modification and any other previously installed modification will introduce no adverse...</td>
<td>06/06/2007</td>
<td>Active</td>
<td>1014423</td>
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<td>UNI AIR ENTREPRISE SASUNI AIR ENTREPRISE</td>
<td>BEECH 300C</td>
<td>Master Data List, Doc. No. MDL.058.07 Revision 0, dated 03-Mar-2007, or later approved revision</td>
<td>Master Data List, Doc. No. MDL.058.07 Revision 0, dated 03-Mar-2007, or later approved revision</td>
<td>30/04/2007</td>
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<td>UNI AIR ENTREPRISE SASUNI AIR ENTREPRISE</td>
<td>BEECH 300C</td>
<td>Dual Installation of Collins TDR-94D with CTL-92E for compliance with P-RNAV</td>
<td>This change is limited to the following serial numbers: B300: S/N FL-111 thru FL-380; and FL-382B300C: S/N FM-9 thru FM-11; and FN-2 and subsequent. This change is limited to the configuration of aircraft/equipment type as listed under Item I - Description Details in the Compliance Design Document...</td>
<td>11/09/2007</td>
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<td>1014825</td>
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<td>JET AVIATION VIENNA GMBH</td>
<td>CESSNA 500</td>
<td>Certification Report C650-AR34-001 - Instructions for continued airworthiness C650-AR34-001-M1C-ICA - AFM Supplement C650-AR34-001-M1A-AMF - Engineering Bulletin EB-C650-AR34-001.</td>
<td>1. Serial number : 650-7001 and on2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot; Compatibility with other aircraft engine configurations shall be determined...</td>
<td>29/05/2007</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.03040</td>
<td>JET AVIATION VIENNA GMBH</td>
<td>650</td>
<td></td>
<td>Upgrade of a MkVI GPWS to a MkVII EGPSW</td>
<td>Certification Report C650-ARS34-002; Instructions for Continued Airworthiness C650-ARS34-002; IFC/AAMF Supplement C650-ARS34-002; IFC/AAMF Rev. 16; Engineering Bulletin EB-C650-ARS-002</td>
<td>Serial number: 650-7001 and on</td>
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<td>29/05/2007</td>
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<td>EASA.A.S.03041</td>
<td>GULFSTREAM AEROSPACE Ltd.</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>Gulfstream GV-X</td>
<td>EASA Certification of a number of FAA Major Alterations installed at original delivery.</td>
<td>Installation in accordance with FAA, modification No. 36701, latest approved revision; Aircraft to be maintained in accordance with Instructions for Continued Airworthiness ref. CE41801A002 Rev. 17/9/07, or later approved revisions. To be operated in accordance with the Cabin Operating Manual...</td>
<td>Aircraft GV-X Serial No. 4004 only.</td>
<td>08/10/2007</td>
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<td>EASA.A.S.03042</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
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<td></td>
<td>550 BRAVO, 560, 560 XL 590, 550, 550S, 560, 560XL</td>
<td>Installation of Honeywell MK VIII EGPSWS.</td>
<td>Master Document List 125706-0109, Issue 1, or later approved revision; Instructions for Continued Airworthiness 125706-0101, Rev 0, dated 02.08.2007, or later approved revision; AFM Supplement 125706-0601, Rev 1, dated 17.08.2007, or later approved revision.</td>
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<td>07/09/2007</td>
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<td>EASA.A.S.03045</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>BOEING B737-800</td>
<td>Installation of a Satcom Antenna for a GSM system on B737-800</td>
<td>HA4891 issue 3, or later EASA approved revision; HA4ACP03 issue 3, or later EASA approved revision; Instruction for Continued Airworthiness HA4DR259 Issue 1, or later EASA approved revision; Weight and Balance report HA4WBB03 Issue 2, or later EASA approved revision.</td>
<td>As in Instructions for Continued Airworthiness HA4DR259 Issue 1 or later EASA approved revision.</td>
<td>16/01/2008</td>
<td>Active</td>
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<td>EASA.A.S.03046</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td></td>
<td>BEECH F90 BEECH C90 BEECH C90</td>
<td>Installation of dual Mode S transponder system (GARMIN GTX-330 or GTX-330D)</td>
<td>Installation in accordance with BCA Master Drawing List BCA95-00, Rev 0, dated 03/02/2007, or later approved revisions. To be operated in accordance with BCA Flight Manual Supplement ref BCA95-05, Rev 1, dated 05/14/2007, or later approved revisions. To be maintained in accordance with BCA Aircraft Ma.</td>
<td></td>
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<td>22/05/2007</td>
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<td>10014331</td>
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<td>EASA.A.S.03048</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td>LEARJET 31A</td>
<td>Navigation – Retrofit of Dual FMC</td>
<td>Compliance Record List CrL 34-0701M, Issue 1 (rev. 0)</td>
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<td>None</td>
<td>02/04/2007</td>
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<td>EASA.A.S.03050</td>
<td>ELECTRONIQUÉ MAINTENANCE AVIATION</td>
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<td>BEECH 200</td>
<td>Installation of Garmin GTX330D Mode S Transponder.</td>
<td>Installation in accordance with EMA STC Data List Doc-No EMA/07/001, BE200/ATC Mode S, Rev 1, or later approved Revisions.To be operated in accordance with EASA Airplane Flight Manual Supplement Doc-No. 05 AFMS S128, Rev 0, or later approved Revisions.To be maintained i.e. EASA Maintenance Manual Doc...</td>
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<td>21/04/2008</td>
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<td>EASA.A.S.03051</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>EASA STC 2004-2474</td>
<td>CESSNA 500</td>
<td>Major Modification: change of a pilot HSI against EHSI ; re-wiring of existing equipment power supply to the emergency bus. Note: This EASA STC supersedes EASA STC 2004-2474.</td>
<td>Description: BCA35 Monopilote Citation I Rev 8. AFM Supplement: SMV BCA35 Rev 5, Master Drawing List: BCA35-00 Rev 5, BCA35-7, Etude des cas de panne, Rev 1.</td>
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<td>02/03/2007</td>
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<td>EASA.A.S.03056</td>
<td>FRANCE AVIATION</td>
<td>BEECH B200</td>
<td>Installation of Honeywell MST-67A Mode S Transponder.</td>
<td>Installation must be in accordance with France Aviation Master Data List Rev 2, dated 13.06.07 or later Approved Revisions. Operation must be in accordance with France Aviation Aircraft Flight Manual Supplement AFM-01-2007-021 Rev 0 dated 16.04.07 or later approved revision. Maintenance must be in acc...</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>18/06/2007</td>
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<td>EASA.A.S.03057</td>
<td>J &amp; C ENGINEERING SERVICES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING B737-300</td>
<td>TCAS II Installation</td>
<td>Limited to installation on Boeing 737-300MSN 28740. Compatibility of this design change with other previously approved modifications must be determined by the installer.</td>
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<td>29/05/2007</td>
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<td>EASA.A.S.03058</td>
<td>J &amp; C ENGINEERING SERVICES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING B737-300</td>
<td>Installation of Dressen Galexy No 1</td>
<td>Limited to installation on Boeing 737-300MSN 28740. Compatibility of this design change with other previously approved modifications must be determined by the installer.</td>
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<td>31/05/2007</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td></td>
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<td>BEECH B200</td>
<td>Installation of one GTX330D and one TDR94.</td>
<td>Installation must be in accordance with S.E.A.M. MDL 1587-07MDLS/SEAM Ed 1, Rev. 0 dated 28.05.07 or later Approved Revisions. Operation must be in accordance with S.E.A.M. Aircraft Flight Manual Supplement for GTX330D at TDR94 Rev. 0 dated 28.03.07 or later approved revisions.</td>
<td>Limited to Beechcraft B200 SN BB-116, F-GDJS. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation must be in accordance with the EASA approved revisions to the Aircraft Flight Manual.</td>
<td>31/05/2007</td>
<td>Active</td>
<td>10179162</td>
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<tr>
<td>EASA.A.S.03060</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td></td>
<td></td>
<td>BOEING 767-306</td>
<td>Interior Changes for the B/E Aerospace First Class Seat Integration in the Forward Passenger Cabin. This modification removes the existing 16 off Weber Business class seats and installs 12 B/E Aerospace First Class Passenger Seats with Ottomans and Sideledges on a Boeing 767-300 for PrivatAir in the...</td>
<td>Modification Summary No. H4A887, Issue 1, dated 22nd May 2007.</td>
<td>1. Instructions for Continued Airworthiness. Ref. No. H4ADR0293. This approval is limited to only the installation made in the Boeing Model 767-306, Serial Number 30393 aircraft. Prior to installation of this modification the installer must determine that the interrelationship between the...</td>
<td>19/06/2007</td>
<td>Active</td>
<td>10144843</td>
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<tr>
<td>EASA.A.S.03065</td>
<td>VITROCET S.p.A.</td>
<td>REV. 1</td>
<td>EASA.A.S. 03065</td>
<td>SA 227</td>
<td>GPWS MKVII/VI replacement with EGPWS MKV/MKV VII Honeywell Note: This STC supersedes EASA.A.S.03065.</td>
<td>AFM Supplement DRW No. 34-40-0706-0850-000 Rev. IR dated 09.02.2007 and subsequent approved revisions. Master Drawing List DRW No. 34-40-0706-0000-000 Rev. IR dated 16.02.2007 and subsequent approved revisions.</td>
<td>This STC is approved only for the product configuration as defined in the approved supplement. Compliance with other aircraft specifications shall be determined by the installer. This Certificate shall remain valid.</td>
<td>07/01/2009</td>
<td>Active</td>
<td>10043124</td>
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<td>EASA.A.S.03067</td>
<td>AVIONITEC AG</td>
<td></td>
<td></td>
<td>PA-31T</td>
<td>Replacing of existing avionics by newer generation avionics in accordance with Unit change overview Nr. 083247-050.</td>
<td>Installation must be in accordance with Avionics Master Document List No. 63247-017 Rev. 1 dated 19 December 2007 and later approved revisions. Operation must be in accordance with Avionics Aircraft Flight Manual Supplement No. 63247-011 and 063247-012 dated 03 October 2007 and 063247-013 dated 26th...</td>
<td>Limited to S/N 31T-7904104</td>
<td>17/01/2008</td>
<td>Active</td>
<td>10014846</td>
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<tr>
<td>EASA.A.S.03069</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td></td>
<td>MYSTERE FALCON 50</td>
<td>Incorporation of Runway Awareness and Advisory System (RAAS) function on EGPWS Honeywell MK V.</td>
<td>• Compliance Checklist DFS 5633-CC1 Issue A1 or later approved revisions. • Master Document List DFS-DE-17-F5-5633Issue A1 or later approved revisions. • Aircraft Flight Manual Supplement DFS N° 54, Rev Original or later approved revisions for those aircraft with EGPWS Mk V with TDR94.</td>
<td>1. This STC is approved only for those Mystere Falcon Starcraft equipped with a Honeywell EGPWS Mk V with Software version 212-212 or later and with Proline Avionic Suite VII and Terrain Display. For additional details please check the DASSAULT Avions S.A.P. TC.</td>
<td>30/06/2007</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.03072</td>
<td>AUSTRIAN AIRLINES AG</td>
<td>REV. 1</td>
<td></td>
<td>A320 AND A321 SERIES</td>
<td>Replace Airshow Digital Interface Unit (DIU) Model 210 by Model 212. This design change comprises the removal of Airshow210 Unit (P/N 910030 or 910040) and installation of Airshow 420 Unit (P/N 921006, 921005 or 921006).</td>
<td>Supplemental Type Definition Report, ref. STD-432F-23-0008-00, dated 03.08.2007 or later EASA approved revisions. Declaration of Compliance, ref. DC- A32F-23-0008-00, dated 02.08.2007</td>
<td>21/01/2008</td>
<td>Active</td>
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<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NA AIC no.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>AIRCRAFT INDUSTRIES, a.s.</td>
<td>CAA C2</td>
<td>STC–010</td>
<td>L 410 UVP-E20</td>
<td>Compilation recorded parameters for FUR BUR 1-2G for compliance with Brazilian regulation RBHA 135.152.</td>
<td>Supplement No. 85 to Flight Manual of L410 UVP-E20 and Supplement No. 108 to Maintenance Manual of L410 UVP-E20</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>22/03/2007</td>
<td>Active</td>
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<td>UNI AIR ENTREPRISE SASUNI AIR ENTREPRISE</td>
<td></td>
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<td>BEECH B300</td>
<td>Installation of a Digital Flight Data Recorder</td>
<td>Master Data List ref. MDL-012-07 revision 1, dated April 20, 2007 or later approved revision</td>
<td>1. This STC is limited to the following S/Ns: B300 S/N FL-111 through FL-380, FL-3829 Beech B300C S/N FM-9 through FM-11, FM-2 and after 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification.</td>
<td>06/06/2007</td>
<td>Active</td>
<td>1014653</td>
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<tr>
<td>AEROSPACE DESIGN AND ENGINEERING CONSULTANTS Ltd.</td>
<td>CAA UK RESPONSIBILITIES</td>
<td>767-300</td>
<td></td>
<td>Introduction of 25X/18Y passenger configuration and associated customisation.</td>
<td>Modification summary (MOD-0187, Revision B, 26/03/2007)</td>
<td>Serial Number 24962 only.</td>
<td>14/06/2007</td>
<td>Active</td>
<td>1014925</td>
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<td>SELL GmbH</td>
<td>REV. 1</td>
<td>BOEING 777-300</td>
<td></td>
<td>Galaxy Reconfiguration and Replacement for Emirates Airlines</td>
<td>Self Type Investigation Document IMDP/03STC-012/STC2N issue dated 3 July 2007</td>
<td>Limited to B777-300 serial numbers: 28860, 28867, 29062, 29063, 29064, 29067, 29395, 29396, 29551, 29559, 29700, 32699, 32700, 32702</td>
<td>31/10/2007</td>
<td>Active</td>
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<tr>
<td>JET AVIATION AG, BASEL</td>
<td>BOMBARDIER BD-700-1A10</td>
<td></td>
<td></td>
<td>W-LAN Activation By installing A sealed dipole antenna in the cabin, a W-LAN ON/OFF switch and new wiring between the router, antenna and the HD-710.</td>
<td>• Declaration of Compliance, Doc. No. BGLG23261-01, Master Data List, Doc. No. BGLG23261-001, Rev. A or latest approved revision. • AFM supplement No. 0733261-011-01, Rev. B or latest approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>06/11/2007</td>
<td>Active</td>
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<tr>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>BOMBARDIER CL-600-2B16 (CL-601) and BOMBARDIER CL-600-2A12 (CL-601) and BOMBARDIER CL-600-1A11 (CL-600)</td>
<td></td>
<td>Honeywell EGPWS MK VII</td>
<td>Modification: 076E-CNAFM: 0716-EAFMS Rev. and N/C dated 16/04/07 or later approved revision MM-0716E-EO</td>
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<td>11/05/2007</td>
<td>Active</td>
<td>1014959</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>BOMBARDIER CL-600-1A11 (CL-600)</td>
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<td></td>
<td>ACSS Mode-S Transponder RZC-850 - EL/ESHS</td>
<td>Modification: 076E-CNAFM: 0716-EAFMS Rev. N/C dated 19/04/07 or later approved revision MM-0716E-EO</td>
<td>With this modification now embodied the aircraft is non EASA compliant.</td>
<td>11/05/2007</td>
<td>Active</td>
<td>1014959</td>
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<td>SOGERMA SERVICES</td>
<td>AIRBUS A300 MODEL 82-1C</td>
<td>Replacement of the obsolete RICA weather radar by a Honeywell RDR-4B colour display radar system with predictive windshear function.</td>
<td>• Modification Approval Sheet, ref. SOG-34-2000 Issue 1, dated 10 May 2001 - Aircraft Flight Manual Supplement, ref. AFM-SOG-40-34-2000 Issue 8</td>
<td></td>
<td></td>
<td>1. This STC is limited to the MSN 0032. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>29/05/2007</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>STC ref.</td>
<td>Model</td>
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<td>Associated Technical Documentation</td>
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<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>AVIONTEC Ltd.</td>
<td>EASA.A.S.03088</td>
<td>M2GU</td>
<td>Installation of replaced Avionics according CP No. 073372</td>
<td>According to Avionitec Master Document List Doc. No. 073372-017 Rev. 1, or later Revision(s) (Note: To be operated in accordance with approved Avionitec AFM Document No. 073372-011. Original issue from 23.05.2008, or later approved Revision(s) to be maintained i.e. Avionitec ICA Document No. 073372-015. Origin: EASA STC No. EASA.A.S.03097 is no longer applicable on EASA STC No. EASA.A.S.03097, is required prior to installation of this STC. 2. The installation of this modification will introduce no adverse effect upon the airworthiness of the product. 1) The installation of this modification by third persons...)</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 1) The installation of this modification by third persons...)</td>
<td>21/11/2008</td>
<td>Active</td>
<td>10116625</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>EASA.A.S.03089</td>
<td>A340-300</td>
<td>Installation of new A340 FINNAIR Cabin and IFE Upgrade</td>
<td>Certification Program ref. doc. CP-A340-07-005, Rev. 3, dated 15th July 2006; Modification File ref. doc. MF-A340-07-005, Rev. 0, dated 17th September 2006; Change Approval Sheet ref. doc. CAS- A340-07-005, dated 17th September 2006; Instruction for Continued Airworthiness, ref. doc. I...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 1) The installation of this modification by third persons...)</td>
<td>18/02/2008</td>
<td>Active</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>EASA.A.S.03090</td>
<td>CL-600-2A12 (CL-601 BOMBBARDIER, CL-600-1A11 (CL-600-2B19)</td>
<td>Dual TAWS Sandel ST3400 - Class A</td>
<td>RUAG Master Document List No. 191D-MOL Revision A or later approved revision RUAG Airplane Flight Manual Supplement No. 191D- AFMS Revision A or later approved revision</td>
<td>Modification Summary Sheet: H4ADR331 Issue 3 or later EASA approved revision Airplane Flight Manual Supplement: H4AFMS018 Rev. IR or later EASA approved revision Additional Instructions for Continued Airworthiness: H4ADR331 or later EASA approved revision...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) The installation of this modification by third persons...)</td>
<td>29/02/2008</td>
<td>Active</td>
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<tr>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>EASA.A.S.03091</td>
<td>BOEING B737-800</td>
<td>Installation of GSM Phone / SATCOM system on B737-800 aircraft.</td>
<td>Modification Summary Sheet: H44/9/1 Issue 3 or later EASA approved revision Airplane Flight Manual Supplement: H4AFMS018 Rev. IR or later EASA approved revision Additional Instructions for Continued Airworthiness: H4ADR331 or later EASA approved revision...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) The installation of this modification by third persons...)</td>
<td>12/12/2008</td>
<td>Active</td>
<td>10114667</td>
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<tr>
<td>JET AVIATION AG, BASEL</td>
<td>EASA.A.S.03092</td>
<td>AIRBUS A319-115</td>
<td>Wireless Local Area Network (WLAN) Activation. The modification includes the following: The antenna will be connected and the WLAN will be activated.</td>
<td>- Master Data List, ref. BA31923261-001, Rev. IR, dated 08.08.07 - Certification Plan, ref. BA31923261-002, Rev. E, dated 31.07.07 - Certification Compliance Sheet, ref. BA31923261-102-01, Rev. A, dated 01.08.07 - Airplane Flight Manual Supplement, ref. BA31923261-201-01, Rev. IR, dated 08.08.07...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 1) The installation of this modification by third persons...)</td>
<td>13/08/2007</td>
<td>Active</td>
<td>10114669</td>
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<tr>
<td>LUFTHANSA TECHNIK AG</td>
<td>EASA.A.S.03094</td>
<td>CL-600-2B19</td>
<td>Cabin Completion - 15 Passengers</td>
<td>- Luftpansa Technik Supplement Type Definition Document No. STD00-07/03M Issue 1 dated 26 February 2008 or later approved revision - Airplane Flight Manual Supplement No. A-00-07/03M-AFM Issue 01 or later approved revision</td>
<td>1. Installation of &quot;Cl-600-2B19 Installation of Structural Provisions for Cabin Completion&quot; approved with EASA STC No. EASA.A.S. 03097, is required prior to installation of RvS STC. 2) The Limitation &quot;Installation of Ferry- Kit MS TC 601R16907&quot; given on EASA STC No. EASA.A.S. 03097 is no longer...</td>
<td>11/03/2008</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.03095</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Cabin Completion</td>
<td>Lufthansa Technik Supplemental Type Definition Document No. STDD00-0707M Issue 01 dated 13/02/08 or later approved revision - Airframe Flight Manual Supplement No. A-00-07069-AM Issue 01 or later approved revision.</td>
<td>1. Installation of &quot;CL-600-2B19 - Installation of structural provisions for cabin completion&quot;, approved with EASA STC No. EASA.A.S. 03097 is required prior to the installation of this STC.2. The limitation &quot;Installation of Ferry-Kit M/S TC 601R16990 (967 [kg]) given on EASA STC No. EASA.A.S. 03097 is no longer valid.&quot;</td>
<td>27/02/2008</td>
<td>Active</td>
<td>1014672</td>
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<tr>
<td>EASA.A.S.03096</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Installation of Structural Provisions for AeroH Antenna System</td>
<td>LHT Declaration of Compliance No. 1613 dated 22 January 2008 - HT Supplemental Type Definition Document No. STDD-0708M dated 22 January 2008</td>
<td>1. For all flights with &quot;Green Aircraft&quot; and &quot;Structural Provisions for AeroH Antenna System&quot; (2,5 [kg]) installed the installation of Ferry-Kit M/S TC 601R16990 (967 [kg]) is necessary.2. With the installation of Ferry-Kit M/S TC 601R16990 (967 [kg]) the aircraft will stay in C.G. envelope 3. Install...</td>
<td>05/02/2008</td>
<td>Active</td>
<td>1014671</td>
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<td>EASA.A.S.03097</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Installation of Structural Provisions for Cabin Completion</td>
<td>LHT Declaration of Compliance No. 1611 dated 29 January 2008 - HT Supplemental Type Definition Document No. STDD-0709M, Issue 1 dated 29 January 2008</td>
<td>11. For all flights with &quot;Green Aircraft&quot; and &quot;Structural Provisions for cabin completion&quot; (293,7 [kg]) installed the installation of Ferry-Kit M/S TC 601R16990 (967 [kg]) is necessary.2. With the installation of Ferry-Kit M/S TC 601R16990 (907 [kg]) the aircraft will stay in C.G. envelope.5. Install...</td>
<td>05/02/2008</td>
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<td>EASA.A.S.03098</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>AIRBUS A330-200</td>
<td>A330 - Installation of Zonal Drying System - The system is installed to get rid of the humidity between the cabin lining and the fuselage</td>
<td>- Supplemental Type Definition Document, Ref. STDD01-079/00M, Issue 01, dated 25 May 2007. Instructions for Continued Airworthiness, Ref. CS 21-0186, Issue 01, dated 17 April 2007.</td>
<td></td>
<td>25/06/2007</td>
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<td>EASA.A.S.03100</td>
<td>UNI AR ENTREPRISE SASUNI AIR ENTERPRISE</td>
<td>LOCKHEED JETSTAR 1329-A</td>
<td>Upgrade of ATC transponders for elementary surveillance (Mode S). To add elementary surveillance functionality in the ATC transponder and replace KFS-578 control box by PS-578 to comply the ECAC airspace requirements in accordance with Master Data List MDL-010-07, dated 11 May, 2007.</td>
<td>Data package containing the following documents, and additional supporting data: MDL-010-07 Master Data List Modification UAE-010-07 DCC-010-07 Compliance Design Document Modification UAE-010-07 UAE-CD-E-0051-07 Ground Test Functional Procedure for a Lockheed JetStar UAE-CD-E-0057-07 Flight Test Proc...</td>
<td>Prior to installation of this modification the installer must determinethat the interrelationship between this modification and any otherpreviously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>06/06/2007</td>
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<td>1014675</td>
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<td>EASA.A.S.03103</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>BOEING 767-300</td>
<td>The cabin reconfiguration of 2876-300 Aircraft with Brisa B10005 seats for TUI ARKELYFLY. This modification removes the existing 260 economy class seats and installs 260 new Brisa B10006 economy class seats in an identical passenger layout.</td>
<td>Modification Summary No. BB0055, Issue 1, dated 18th May 2007.</td>
<td>Continued Airworthiness Manual, Ref. No. ML3263, Rev. A. This approval is limited to only the installation made in the Boeing Models B767-300, S/N 24477, S/N 24847, S/N 24848 aircraft.</td>
<td>14/06/2007</td>
<td>Active</td>
<td>1014676</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<td>VITROCISET S.p.A.</td>
<td>EASA A.S. 03106</td>
<td>Installation of Honeywell MVI-1 EOPWS. Note: This STC supercedes EASA.E.03106 Revision 1.</td>
<td>AFM Supplement DRW No. 34-97-0407-0600-000 Rev. IR and subsequent approved revisions. Master Drawing List DRW No. 34-97-0407-0000-000 Rev. IR dated 2 July 2007 and subsequent approved revisions.</td>
<td>1. Last B.M. No. 15-07 must be applied before the installation. 2. Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification has no adverse effect upon the airworthiness of the product.</td>
<td>07/01/2009</td>
<td>Active</td>
<td>12014737</td>
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<tr>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>REV. 1</td>
<td>CESSNA 650 CITATION VII CESSNA 650 CITATION VI CESSNA 650 CITATION III</td>
<td>Mode S upgrade of existing Honeywell transponder system RCZ-833 to comply with ICAO requirements for Mode S ELS and EMS. Minor changes to EASA.A.S. 03107 AFM Supplements for different Downlinked Aircraft Parameters: Mode S upgrade of existing Honeywell transponder system RCZ-833 to comply with ICAO requirements.</td>
<td>Installation in accordance with Scandinavian Avionics Flight Manual Supplement Ret 2037868-AM-A Revision 1, 2037868-AM-B as required, both at issue 1, date.</td>
<td>20/11/2007</td>
<td>Active</td>
<td>12014791</td>
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<tr>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>CESSNA CITATION CS 551 CESSNA CITATION 551</td>
<td>Installation of Dual Mode S XPDR MST-47A Model 501 S/N 0001 and on (Citation II-SP)</td>
<td>Master Doc. List: 204A-MMIL Engineering Order: 204A-EO - EASA approved RUAG AFM-S Doc. No. 204A-AMMS. Original Issue, or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>21/06/2007</td>
<td>Active</td>
<td>12014801</td>
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<tr>
<td>AEROSPACE DESIGN AND ENGINEERING CONSULTANTS Ltd.</td>
<td>737-300</td>
<td>Introduction of 138 Passenger Configuration and Associated Customizations</td>
<td>Modification Summary MOD-0217 Rev A, dated 02 July 2007, or later EASA approved revision. Aircraft Flight Manual in accordance with paragraph 3.4 of Modification Summary MOD-0217 Rev A, dated 02 July 2007, or later EASA approved revision. Weigh &amp; Balance Manual in accordance with paragraph 3.7.4...</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>20/09/2007</td>
<td>Active</td>
<td>12014881</td>
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<tr>
<td>H4 AEROSPACE (UK) Ltd</td>
<td>BOEING 737-500</td>
<td>Implementation of interior changes associated with new cabin configuration from 124YC to 129YC.</td>
<td>Most Summary H44876 Iss. 204A Instructions for Continued Airworthiness Doc. H44DR383 Iss 1</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>10/06/2008</td>
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<td>12014882</td>
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<td>EASA STC no.</td>
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<td>Foreign/NAAs (STC ref.)</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>EASA.A.S.03114</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>0</td>
<td>CESSNA 421</td>
<td>Installation of 4-blade MT-Proppell Model MTV-16-1-E-C-F/CF225-37a propellers on Cesna 421 C.</td>
<td>Installation Instructions Document E-1550MT-Proppell AFM-Sdoc. No. E-1554. Original issue from 23 November 2007, or later EASA approved Revision. MT-Proppell Instructions for ConAW Doc No. E-1388. Original issue from 23 November 2007, or later approved Revision.</td>
<td>Compliance has also been shown for all combinations (and/or) of following FAA STCs: SA438NW, SA5193NM, SA8352SW, SA00015SE, SA9811SW</td>
<td>18/12/2007</td>
</tr>
<tr>
<td>EASA.A.S.03117</td>
<td>JET AVIATION AG, BASEL</td>
<td>0</td>
<td>AIRBUS A319-115</td>
<td>Water Tanks Installation The modification includes the following installation of two additional water tanks in the AFT lower lobe area between frame FR62 and frame FR65 on the LH and RH side.</td>
<td>• Master Data List, ref. BA31938121-001, Rev. IR, dated 10.09.07. Certification Plan, ref. CP-BA31938121, Rev. C, dated 16.08.07. Certification Compliance Sheet, ref. BA31938121-102-01, Rev. C, dated 16.08.07. Instruction for Continued Airworthiness, ref. BA31938121-150-01, Rev. IR, date.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>14/06/2007</td>
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<td>EASA.A.S.03121</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>0</td>
<td>BOMBARDIER CL-600-2A12 (CL-601</td>
<td>EASA recertification of Cabin Modifications</td>
<td>RUAG Master Document List No. 191A-MDL Revision A or later approved revision.</td>
<td>29/02/2008</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.03122</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>0</td>
<td>BOMBARDIER CL-600-2A12 (CL-601</td>
<td>EASA recertification of Avionics Modification</td>
<td>RUAG Master Document List No. 191B-MLD Initial Issue or later approved revision RUAG Airplane Flight Manual Supplements No. 191B-AMFS-TCA, No. 191B-AMFS-FLM, No. 191B-AMFS-COM, No. 191B-AMFS-HFN, No. 191B-AMFS-CVR No. 191B-AMFS-DFD or initial issue or later approved revision.</td>
<td>29/02/2008</td>
<td>Active</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
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<td>Associated Technical Documentation</td>
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<tr>
<td>EASA.A.S.03123</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>REV 1</td>
<td></td>
<td>AIRBUS A319-100</td>
<td>PTU and/or Stretcher installation on A319-100 (Optional replacement of seats by up to 3 Passenger Transport Units (PTUs) and/or up to 4 stretchers.)</td>
<td>- Supplemental Type Definition Document, ref.: Doc. No. ASTM2017-0710M, Issue 1, dated 20.09.07, or later approved/revision(s)- Aircraft Flight Manual Supplement, ref. Doc. No. A-00-0714-04M, dated 20.09.07 or later approved revisions- Instruction for Continued Airworthiness, ref. Doc. A-...</td>
<td>* Detailed visual inspection of the Oxygen Discharge Plug at an interval of 750 F(C). (C check.) as reported in the document Instruction for Continued Airworthiness, ref. Doc. A-00-07141ICA Issue 03, dated 19.09.07</td>
</tr>
<tr>
<td>EASA.A.S.03124</td>
<td>AIR ALPHA MAINTENANCE</td>
<td>200</td>
<td></td>
<td>replacing transponders with Rockwell Collins TDR-94 or TDR-94D Mode 5 systems for ELTs and EHS. Installation of additional Antenna(s) CI110-660 or DIN 150-A.</td>
<td>Air Alpha EO-B200-34-104AFM Supplement FMS.EO-B200-34-104R1</td>
<td></td>
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<tr>
<td>EASA.A.S.03125</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td></td>
<td></td>
<td>DOERNIER 228-212</td>
<td>Adjustment of Special Fatigue Life Mode: Dornier 228-212, SN 8181 and 8183</td>
<td>Supplemental Time Limits/Maintenance Checks Manual for aircraft S/N 8181TM- STLMCM-8181-090707 Issue July 09/2007 Supplemental Time Limits/Maintenance Checks Manual for aircraft S/N 8183TM- STLMCM-8183-090707 Issue July 09/2007. 1. This STC is only applicable to aircraft S/N: 8181, 8183.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of aircraft.</td>
<td>08/10/2007</td>
</tr>
<tr>
<td>EASA.A.S.03126</td>
<td>AIRCRAFT INDUSTRIES, a.s.</td>
<td></td>
<td></td>
<td>Z-37A</td>
<td>Implementation of VFR-NIGHT kind of operation - installation of additional equipment ensuring the extension of currently approved kinds of operation to VFR-NIGHT kind of operation.</td>
<td>Supplement No. 2 to Flight Manual of Z-37A s/n 23-08.</td>
<td>Applicable for s/n 23-08 only.</td>
</tr>
<tr>
<td>EASA.A.S.03128</td>
<td>SABENA TECHNICS BGC</td>
<td></td>
<td></td>
<td>777-200, 777-300</td>
<td>Installation of Aeronautical GSM connectivity system.</td>
<td>Master Data List 0333-01-A-2103-R00 Certification Plan 0333-01-A-2101-R00 Compliance Record 0333-01-A-2104-R00 Aircraft Flight Manual Supplement 0333-01-A-2301-R00 Aircraft Flight Manual Supplement 0333-01-B-2301-R00 for continued Airworthiness Aircraft Flight Manual Supplement 0333-01-A-2211-R00</td>
<td>The applicability of this STC is limited to any Boeing 777-200 or B777-300 shown to be identical to Boeing 777-200, M9SN22747 or Boeing 777-300 M9SN 29083, through the DOA's process as defined within document 0001-01-B-0914 (Classification of Modification Change procedure for GSM connectivity System)...</td>
</tr>
<tr>
<td>EASA.A.S.03129</td>
<td>SOGERMA SERVICES</td>
<td>A310-325</td>
<td></td>
<td>Installation of ATC mode S Enhanced Surveillance System (EHSS) This Supplement Type Certificate consists in the installation of new ATC EHS in order to be compliant with the new ATC mode SEnhanced Surveillance regulation.</td>
<td>Modification Approval Sheet ref.: SOG-234-2005 issue 1, dated 13 July 26/074AFM Supplement, ref.: AFM-SOG-A1-34-2005 Issue E, dated 02 May 26/07Instruction for operation Master Minimum Equipment List Supplement</td>
<td>This STC is limited to A310-325 aircraft model: M9SN 636-M9SN 6442. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previous installed modification will introduce no adverse effect upon the airworthiness...</td>
<td>19/07/2007</td>
</tr>
<tr>
<td>EASA.A.S.03130</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITIES</td>
<td>737- 600, 700, 800, 900</td>
<td>737-200, 300, 400, 500,</td>
<td>Installation of Cabin VU-123 Cabin Surveillance System.</td>
<td>MDL S21, MDL-035 Issue 1, dated 21st January 2008 or later EASA approved revisions.</td>
<td></td>
</tr>
<tr>
<td>EASA.A.S.03135</td>
<td>OCEANA LETEOKA SERVISNI A.S.</td>
<td>CESSNA 172S</td>
<td></td>
<td>CESSNA 172S</td>
<td>Installation of the DME KN 62A, ADF KR-87 and XPRN KT-73</td>
<td>CIL.-CIL-172S-AC152-S14, Dated 27. August 2007CIL.-CIL-172S-AC152-S24, Dated 27. August 2007CIL.-CIL-172S-AC152-S34, Dated 27. August 2007CIL.-CIL-172S-AC152-S44, Dated 27. August 200726/07CIL.-CIL-172S-AC152-S44</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previous installed modification will introduce no adverse effect upon the airworthiness...</td>
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<tr>
<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
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<td>STC A.S.03139</td>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td></td>
<td>GULFSTREAM II</td>
<td>Super FMS Installation by Removing the existing dual UNS-1m Flight Management System (FMS) and installing a new dual UNS-1L Super Flight Management System (FMS)a DTU-100 Data Transfer Unit.</td>
<td>Master Data List, Doc No. B7AA4038-002. Rev A or latest approved revision</td>
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<td>STC A.S.03140</td>
<td>ACI SYSTEMS</td>
<td></td>
<td></td>
<td>REIMS AVIATION CESSNA F 406</td>
<td>Active</td>
<td>Active</td>
<td>08/10/2007</td>
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<td>STC A.S.03151</td>
<td>E.I.S.A.</td>
<td></td>
<td></td>
<td>FALCON 10</td>
<td>MODE S TRANSPONDER WITH ENHANCED SURVEILLANCE CAPACITY.</td>
<td>Active</td>
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<td>STC A.S.03153</td>
<td>TATENHILL AVIATION LTD</td>
<td></td>
<td></td>
<td>CESSNA 500</td>
<td>Approval of a Mode S EHS ATC System.</td>
<td>Tatenhill Aviation Limited Change TAL-TAD 020/075Flight Manual Supplement FMU2007 Revision A or later EASA approved revision</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
</tr>
<tr>
<td>STC A.S.03156</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CO RPRORATION AG (AIEC)</td>
<td>REV 1</td>
<td></td>
<td>CESSNA 550 CITATION II</td>
<td>Honeywell ART.3000 Weather Radar Transceiver/antenna installation interfacing with KMD 850 MFD, to improve reliability and to add display of terrain and traffic graphics.</td>
<td>Master Data list 679-049,AFM Supplement 679-071,ICA 679-085 ICA.</td>
<td></td>
</tr>
<tr>
<td>STC A.S.03157</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CO RPRORATION AG (AIEC)</td>
<td>REV 1</td>
<td></td>
<td>CESSNA 550 CITATION II</td>
<td>Honeywell EGPRS Mark VIII System Installation interfacing with KMD 650 MFD, as a replacement of the existing GPS/M.</td>
<td>Master Data list 681-049,AFM Supplement 681-071,ICA 681-085 ICA, Rev. A.</td>
<td></td>
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<tr>
<td>STC A.S.03159</td>
<td>FRANCE AVIATION</td>
<td></td>
<td></td>
<td>PIPER PA-32</td>
<td>Rudder Hand Control installation for lower Limbs. The &quot;SD1&quot; system consists in a control for rudder that is handled by the pilot while seated in the LH (Left Hand) station. This &quot;SD1&quot; RHC system is mounted on existing manufacturing rudder system. The system can be easily removed (four screws).</td>
<td>Master Data List FA 25-2007 rev. 1 dated 17/07/2007 Flight Manual Supplement AFM-25-2007-001 rev. 1 dated 17/07/2007 Instructions for Continued Airworthiness ICA-25-2007-001 rev. 0 dated 15/06/2007.</td>
<td></td>
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<tr>
<td>STC A.S.03160</td>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td></td>
<td>GV</td>
<td>Installation of bulkheads and door in the mid cabin.</td>
<td>Flight Manual Supplement doc. no. GVCS002-0201-02.</td>
<td>1. S/N 660.2. Prior to installation of this modification the installer must determine that the interrelationships between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modificatio...</td>
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<tr>
<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
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<td>Associated Technical Documentation</td>
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<td>EASA.A.S.03162</td>
<td>AVIONTEC LTD</td>
<td></td>
<td></td>
<td>PA-24R-2017 (TURBO ARROW III)</td>
<td>Replacing of existing avionics by newer generation of avionics in accordance with Unit change overview No. 073528-050, Rev 0, dated 11. 06. 07.</td>
<td>Installation must be in accordance with Aviontec Master Document List No. 073528-017 Rev.0 dated 28. Dec. 2007 or later Approved Revisions. Operation must be in accordance with Aviontec Airplane Flight Manual Supplement No. 073528-011 dated 06. July. 2007 or later approved revision. Maintenance must...</td>
<td>Limited to SN T21059022.</td>
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<tr>
<td>EASA.A.S.03194</td>
<td>STC TWENTY ONE LIMITED</td>
<td></td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BOMBARDIER CL-600-2B16 (604)</td>
<td>Introduction of Passenger Cabin Interior.</td>
<td>STC 21 Master Document List No. S21 MDS-0300 at Issue 1 dated 19 August 2008 or later approved revision. STC 21 Flight Manual Supplement No. S21. FMS-0041 at Issue 1 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
</tr>
<tr>
<td>EASA.A.S.03165</td>
<td>AIM AVIATION (JECCO) LIMITED</td>
<td></td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>747-400</td>
<td>Virgin Atlantic Airways B747-400 CoCo Reconfiguration.</td>
<td>Major change MM373-07 Issue A and Appendix 1 Issue C, and, Master Drawing List DR1281-07, Issue 2, or later approved revisions.</td>
<td>The following EASA approved STC's must be installed concurrently with this design change: EASA.A.S.02860EASA.IM.A.S.02572</td>
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<tr>
<td>EASA.A.S.03167</td>
<td>NSE</td>
<td></td>
<td>177RG</td>
<td>STC 42/2007 - Modification of radio electric installation for B-RNAV certification with AP coupling.</td>
<td>Airplane Flight Manual Supplement Ed1 Rev1, dated 20 September, 2007.</td>
<td>Aircraft must be equipped with HSI KIS33A and S-TEC 60-2 autopilot.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the...</td>
<td>08/02/2008</td>
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<td>EASA.A.S.03168</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td></td>
<td>A300F4-606ST</td>
<td>Implementation of Automatic Dependant Surveillance – Broadcast in non Radar Airspace capability (ADS-B NRA) on A300-606ST. This STC provides surveillance services in non-radar airspaces using ADS-B Out. ADS-B NRA is part of the Eurocontrol ‘Cascades’ project.</td>
<td>- Change Approval Sheet, ref: ST34MOD030 issue 1 (Any later approved revision). - AFM Supplement ref: ST34MOD030, dated 12 November 2008.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>22/12/2008</td>
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<tr>
<td>EASA.A.S.03169</td>
<td>CESSNA DUSSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td></td>
<td>CESSNA 650 CITATION III/IV</td>
<td>Installation of Honeywell CAS-67A and RCC-833 Mode S transponders.</td>
<td>1. Installed in accordance with Jet Aviation Installation Procedure defined in Jet Aviation Order document ref. 2007-14-1,2, Operated in accordance with Aircraft Flight Manual Supplement (AFMS) ref. 2007-14-6.3, Instructions for continued airworthiness defined in Jet Aviation Document ref. M...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>01/02/2008</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NA A STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>EASA.A.S.03170</td>
<td>TATENHILL AVIATION Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>CESSNA 500</td>
<td>Approval of a Mode S EHS ATC System.</td>
<td>Tatenhill Aviation Limited</td>
<td>Changes TAL-TAD 01907 Flight Manual Supplement FMS.01907 Revision A or later EASA approved revision</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.A.S.03176</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td></td>
<td>BEECH 200</td>
<td>Installation of a support base for a stretcher.</td>
<td>Installation in accordance with SEGA Modification ref A-25-02-07-TwinJet, latest approved revision. Operation in accordance with SEGA Supplement Manuel de Vol Beech 200 ref FAM No A-25-02-07-TwinJet, dated May 2007, or later approved revision.</td>
<td></td>
<td>08/10/2007</td>
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<tr>
<td>EASA.A.S.03177</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td></td>
<td>BEECH 90</td>
<td>Installation of a support base for a stretcher.</td>
<td>Installation in accordance with SEGA Modification ref A-25-02-07-TwinJet, latest approved revision. Operation in accordance with SEGA Supplement Manuel de Vol Beech 90 ref FAM No A-25-02-07-TwinJet, dated May 2007, or later approved revision.</td>
<td></td>
<td>08/10/2007</td>
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<tr>
<td>EASA.A.S.03181</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd.</td>
<td></td>
<td>BAE 146-200</td>
<td>SIR Mode S Enhanced Surveillance</td>
<td>Mod.No.BAC/146/100 Issue 1AFM FMS/486/42</td>
<td></td>
<td>16/06/2007</td>
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<td>EASA STC no.</td>
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<td>Model</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>CAS-67A</td>
<td>Cessna</td>
<td>C540</td>
<td>Upgrade to Garmin GTX330D diversity Mode S Transponder.</td>
<td>Installation must be in accordance with EIS Aircraft GmbH Master Document List MDL 96C21002 Issue 06, dated 20.08.07 earlier Approved revisions. Operation must be in accordance with EIS Aircraft GmbH Flight Manual Supplement 96C21002-91501Issue 02, dated 20.08.07 or later approved revision. Maintains...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>21/08/2007</td>
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<td>A.S.3.0187</td>
<td>Avionics Integration &amp; Engineering Corporation AG (AIEC)</td>
<td>BAE.125 SERIES 800</td>
<td>Upgrade of EGPWS MK V to RAAS.</td>
<td>EASA approved Master Data List. Doc. No. 687-049, Revision B, dated 18-Dec-2007, or later approved revision. EASA approved Arlyeine Flight Manual Supplement, Doc. No. 687-071, Revision A, dated 22-Oct-2007, or later approved revision.</td>
<td>This STC is limited to the configuration of aircraft / equipment type as listed under Item 2.1 Limitation, in Engineering Order, Doc. No. 687 - 042, Revision B, dated 18-Dec-2007, or later approved revision.</td>
<td>10/01/2008</td>
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<td>A.S.3.0189</td>
<td>Luftansa Technik AG</td>
<td>AIRBUS solt 12</td>
<td>Installation of a Honeywell Bendix/King Radio KTR 909 System.</td>
<td>• Supplemental Type Definition Document, ref. Doc. No. STDG21-0714M, Issue 09, dated 06 April 2009, or later approved revision. • Aeroplane Flight Manual Supplement, A-00-070703-AM revision 5, dated 30. September 2008, or later approved revision. • Instructions for Continued Airworthiness, ref...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by third persons...</td>
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<td>Description</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.A.S.03202</td>
<td>S.E.A.M. SERVICE ELECTRONIC AVIATION MAINES.A.R.L.</td>
<td></td>
<td>PIPER PA-28R-201T</td>
<td>Installation modification B- RNAV Installation of annunciation unit MFD Continuity MD 41-1484.</td>
<td>1561-07, Rev. 1</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>07/10/2008</td>
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<td>EASA.A.S.03205</td>
<td>KUERZI AVIATION AG</td>
<td></td>
<td>CESSNA 340A</td>
<td>Installation of a Traffic Advisory System By installing a SKY497 Skywatch Traffic Advisory System with appropriate antennas. Traffic information will be displayed on existing systems.</td>
<td>Master Document List, Doc. No. MD-200717371, Declaration of Compliance, Doc. No. DOC-200717371/AFMS, Doc. No. FMS-200717371, Rev. NIC latest approved revision.</td>
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<td>28/06/2007</td>
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<td>EASA.A.S.03207</td>
<td>AIR ALLIANCE GmbH</td>
<td></td>
<td>PA-42 (CHEYENNE III) PA-42-720 (CHEYENNE IIIA) PA-42-1000 (CHEYENNE 400LS)</td>
<td>Installation of TCAS 9000 BX.</td>
<td>Installation Instruction 2007_001_0965_098 Rev. 04 Alliance - Einbau Avionics TX 620 (90088K) TCAS System AFM supplement.</td>
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<td>06/12/2007</td>
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<td>EASA.A.S.03227</td>
<td>SCANDINAVIAN AVIONICS DESIGN A/S</td>
<td></td>
<td>MD-90 MD-88 DC-9-87 (MD-87) DC-9-83 DC-9-82 DC-9-81 (MD-81) DC-9-50 SERIES DC-9-40 SERIES DC-9-30 SERIES DC-9-20 SERIES DC-9-10 SERIES</td>
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<td>EASA.A.S.03230</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CO RPRORATION AG (AIEC)</td>
<td>10043129</td>
<td></td>
<td>BAE 125 -400 SERIES</td>
<td>Modification of Collins Transponders to achieve Enhanced Surveillance Capability.</td>
<td>Installation in accordance with AIEC Engineering Order 688-002 latest approved revision. Aircraft to be operated in accordance with AIEC Aircraft Flight Manual Supplement, Document ref 688-071 latest approved revision. Aircraft to be maintained in accordance with AIEC instructions for Continued Airworthiness.</td>
<td>None.</td>
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<td>EASA.A.S.03231</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>10014965</td>
<td></td>
<td>65-A90, C90</td>
<td>Installation of Air Medical Interior “Quick Conversion”</td>
<td>Beech 65-A90, Beech C90, P/N 65-A90-25-20-15.005.</td>
<td>• AFM Beech 90 Conversion, Air Ambulance Equipment Beech65/PN 65-A90-25-20-15.005, AFM 46 • Component Installation, Operation Manual, Illustrated Part Catalog Aircraft 65-A90 Air Ambulance Equipment 90 P/N 90-25-20-15.005, CMM- Beech-AAT63. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.A.S.03232</td>
<td>RUECKER AEROSPACE GmbH</td>
<td>10014969</td>
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<td>DORNIER 328-100</td>
<td>Installation of x 230VAC Euro standard power socket both cockpit sidewalls.</td>
<td>MASTER Document List MDL-244-0200-000 Compliance document F_DO-019/ B of 21 March 2007 Change Description Numbered 6221070009</td>
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<td>EASA.A.S.03233</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>10014964</td>
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<td>35A</td>
<td>Dual VHF NAV VR-30/GME-40 replacement by VR-2GME-42 with Dual CTL-32 installation (Collins).</td>
<td>AFM Supplement AG882-071, rev. A (or later approved revisions).</td>
<td>N/A</td>
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<td>EASA.A.S.03238</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>10014968</td>
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<td>CESSNA 551</td>
<td>Dual installation of Transponder Mode S System, MST STA with PS 550 or PS 578A Mode S Control Panel, with ELTs and EHS capability.</td>
<td>EASA approved Engineering Order, Doc. No. 2041564-E0, Issue 2, dated 13-Aug-2007 or later approved issue EASA approved Aircraft Flight Manual Supplement, Doc. No. 2041564-AM-A, Issue 1, dated 13-Aug-2007 or later approved issue for installation with PS 550 Control Panel, EU SEASA approved Aircraft.</td>
<td>This change is limited to the configuration aircraft / equipment type as listed under Item 4. in EASA approved Engineering Order, Doc. No. 2041564-E0, Issue 2, dated 13-Aug-2007 or later approved issue</td>
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<td>EASA.A.S.03239</td>
<td>FR AVIATION Ltd D/B/A COBHAM AVIATION SERVICES UK</td>
<td>10014969</td>
<td></td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BEECH B200</td>
<td>RNA V GNSS Non-Precision Approach.</td>
<td>FR Aviation Design Change DC-O 119 FR Aviation AFM Supplements FRAHFM-080 Revision 1 dated 23 August 2007 and FRAHFM-007 Revision 1 dated 23 August 2007. Prior to installation of this modification the installer must determinethat the relationships between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is sub...</td>
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<td>EASA.A.S.03247</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>10014970</td>
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<td>35A</td>
<td>Enhanced Surveillance Mode S Transponder modification (Collins).</td>
<td>AFM Supplement AG883-071, rev. A (or later approved revisions).</td>
<td>N/A</td>
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<td>EASA.A.S.03248</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
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<td>35A</td>
<td>Centre Pedestal Enlargement / Nose Compartment Rearrangement.</td>
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<td>EASA.A.S.03253</td>
<td>LUITHANSA TECHNOK AG</td>
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<td>AIRBUS A340-212</td>
<td>WLAN Installation Design and implementation of Wireless Local Area Network System</td>
<td>Supplemental Type Definition Document, ref. Doc. No. STD23-07/09M, Issue 1, dated 10-10-07 or later approved revisions</td>
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<td>EASA.A.S.03258</td>
<td>JET AVIATION AG, BASEL</td>
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<td>BOEING 737-700 IGW</td>
<td>FMS and DME installation</td>
<td>Master Document List, Doc. No. MDS-200717475, Rev. N/C or latest approved Revision, AFM, Doc. No. FMS-200717475, Rev. N/C or latest approved Revision, Instruction for Continued Airworthiness, Doc. No. ICA-200717475, Rev. N/C or latest approved Revision.</td>
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<td>EASA.A.S.03265</td>
<td>JET AVIATION AG, BASEL</td>
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<td>MYSTERE FALCON 50</td>
<td>Tripe FMS Upgrade to UNS-1F</td>
<td>Master Data List BOHT039-002 Rev. E, dated 07th December 2007.</td>
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<td>EASA.A.S.03267</td>
<td>JET AVIATION AG, BASEL</td>
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<td>FALCON MF 50</td>
<td>MYSTERE</td>
<td>Master Data List BOHT039-001 Rev. D Operation Instructions BOHT039-202-11 Rev. D Instructions for Continued Airworthiness BOHT039-100-11 Revision AE Engineering Order BOHT039-250000 Rev. AO any later approved revisions of these documents.</td>
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## List of EASA Supplemental Type Certificates

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<th>System Record</th>
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<tr>
<td>EASA.A.S.03272</td>
<td>EADS ELBE FLUGZEUGWERK E GmbH</td>
<td>AIRBUS A310-304</td>
<td>A310-304 Cabin Reconfiguration/Transfer of the STC holder, from ASL Aircraft Services Lemwerder GmbH to EADS Elbe Flugzeugwerke GmbH/Former LBA STC ref TA0721/Cabin Reconfiguration from 265 seats to 225 seats (118 Business Class + 207 Economy Class).</td>
<td>Certification Report, ref: ASL CR 310-25-0001, dated 11 April 2003</td>
<td>1. This STC is limited to A310-304 aircraft model MSN 445.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the prod...</td>
<td>08/10/2007</td>
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<td>10014922</td>
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<td>EASA.A.S.03273</td>
<td>EADS ELBE FLUGZEUGWERK E GmbH</td>
<td>AIRBUS A310-304</td>
<td>A310-304 Floor Path Marking System Installation/Transfer of the STC holder, from ASL Aircraft Services Lemwerder GmbH to EADS Elbe Flugzeugwerke GmbH/Former LBA STC ref TA0719/The modification related to this supplement concerns the installation of a photoluminescent emergency escape path marking.</td>
<td>Certification Report, ref: ASL CR 310-33-0001, dated 10 April 2003/Aircraft Flight Manual Supplement, ref: AFMS310330001, dated 09 March 2003</td>
<td>1. This STC is limited to A310-304 aircraft model MSN 445.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the prod...</td>
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<td>EASA.A.S.03274</td>
<td>EADS ELBE FLUGZEUGWERK E GmbH</td>
<td>AIRBUS A320-214</td>
<td>A320-214 Cabin Reconfiguration/Transfer of the STC holder, from ASL Aircraft Services Lemwerder GmbH to EADS Elbe Flugzeugwerke GmbH/Former LBA STC ref TA0717 issue 2/Cabin Reconfiguration from 156 seats (90 Business Class + 66 Economic Class) to 180 seats (Economy Class).</td>
<td>Certification Report, ref: ASL CR 320-25-0001/Revision B, dated 20 May 2003</td>
<td>1. This STC is limited to A320-214 aircraft model MSN 1081 and MSN 1370.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the prod...</td>
<td>08/10/2007</td>
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<td>EASA.A.S.03275</td>
<td>AEROLITE AG</td>
<td>MD-11</td>
<td>Modification of two SDMA Mono FIC Seat Type 17111(U)-100/106 detailed in Luftansa Supplemental Type Definition Document STD 26-05-02 R2 and in accordance with Aerolite BucherMaster Document List MDL-06-015 Rev. 8 for installation in MD-11 MSN48533.</td>
<td>MLD-06-015 Rev.8 Master Document List/Installation Manual VVP Seat Installation</td>
<td>It is the responsibility of the installer to make sure that this installation is certified for the use in effective aircraft.Compliance with the following Cabin Safety requirements is not covered by this approval 1. FAR/JAR 25.791 Passenger information signs and placards 2. FAR/JAR 25.812(a)(1) Signs and placards</td>
<td>16/11/2007</td>
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<td>EASA STC No.</td>
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<td>EASA.A.S.03281</td>
<td>AIR ALLIANCE GmbH</td>
<td>PA-31T (CHEYENNE/ CHEYENNE II)</td>
<td>Installation of Avionic Upgrade according to MDL 2007_019_2029AA Form 1 Rev. 0</td>
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<td>10/01/2008</td>
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<td>EASA.A.S.03284</td>
<td>CESKA LETECKA SERVISNI A.S.</td>
<td>LETECHE ZAVODY L-410-UVP-E</td>
<td>Installation of ACAS II system with XPDOR (MBT-67A) and ELT defined in document CLS-06-L410-AC125, revision 02 dated 04.11.2005</td>
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<td>EASA.A.S.03288</td>
<td>AIR SUPPORT INTERNATIONAL S.r.l.</td>
<td>PA-31T (CHEYENNE/ CHEYENNE II)</td>
<td>ASI BMA 048B6 “Installation of Garmin GNS 430 &amp; GNS 530, Goodrich WX 500, Garmin MX 20 and Accessories”</td>
<td>AFM Supplement No. BMA 048B6</td>
<td>1. Prior to installation of this modification the installers must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.A.S.03292</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>228-202, -212, 228-100, -101, -200, -201</td>
<td>Installation of dual Garmin GB 533/4 AW NAV/COM/MAP/SPS and dual sandel SN4500 E HS</td>
<td>POH supplement AS412, Edition 1 Engineering Order: EO-226-234, Rev. 0 Continued Airworthiness TM-TL2CMC-090305- ALL Chapter 05-21-10 Vertical approach navigation based on GPS/WAAS is only on board full installation approved in VMC Restricted to aircraft equipped with Honeywell King Silver Crown Series</td>
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<td>EASA.A.S.03293</td>
<td>ACI SYSTEMES</td>
<td>BEECH 90</td>
<td>GNS 430 and B-RNAV capability on Beach 90</td>
<td>AFM Supplement ACI-09- AFM-8407-Rev0</td>
<td>N/A</td>
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<td>EASA.A.S.03297</td>
<td>SLC TWENTY ONE LIMITED</td>
<td>CA/UV RESPONSIBILITY</td>
<td>BOEING 767-300 BOEING 767-200</td>
<td>STC 21 MSL 2208 MDL-3093, Issue 1, 29/02/2007</td>
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<td>20/05/2008</td>
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<td>EASA.A.S.03299</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD.</td>
<td>Airbus 1-11</td>
<td>TAWS Installation.</td>
<td>Flight Manual Supplement FM.5254/4 Rev/Modification Document BAC/111008 issues 4 and 5</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>30/09/2008</td>
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<td>EASA.A.S.03300</td>
<td>JET AVIATION AG, BASEL</td>
<td>10015000</td>
<td>BOEING 767</td>
<td>Activation of RAAS.</td>
<td>Master Document List Doc. No.: BB767346-001 AFM supplement Doc. No.: B7673461-210-01</td>
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<td>EASA.A.S.03301</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD.</td>
<td>10015005</td>
<td>BAE 146</td>
<td>Installation of VIP LOPA (Inc. Club 4 Layout).</td>
<td>Modification BAC/146/10Instructions for Continued Airworthiness ICA/033/525</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.A.S.03303</td>
<td>ACI SYSTEMES</td>
<td>10015007</td>
<td>FALCON 900 FALCON 50</td>
<td>Falcon 50 and F900 EHS Mode S Capability</td>
<td>AFM Supplement ACI-05-AFM-M-033-RevACI-05-TO-M-033-Rv0</td>
<td>Applicability for Falcon 50 and F900 which are already Mode A, C and ELB Mode S certified. ACAS certified and equipped with a dual transponder system.</td>
<td>29/10/2007</td>
<td>Orphaned</td>
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<td>EASA.A.S.03305</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>10015008</td>
<td>CESSNA 650 CITATION VII CESSNA 650 CITATION VI CESSNA 650 CITATION III</td>
<td>FMS Upgrade to Dual Honeywell NZ-2000/PRNAV</td>
<td>Aircraft Flight Manual Supplement, Doc. 223A-AFM, Rev. IR or later approved revision.Continued Airworthiness, Doc. 223A-CAPW, Rev. IR or later approved revision.</td>
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<td>EASA.A.S.03306</td>
<td>DASSAULT FALCON SERVICE</td>
<td>10015009</td>
<td>FALCON 2000EX FALCON 2000</td>
<td>F2000 / F2000EX Incorporation of Runway Awareness and Advisory System (RAAS) Function on EGPWS Honeywell Mk V.</td>
<td>1. DFS Master Document List DFS-DE-17-FD-D100 Issue A1 or later approved revisions.2. DFS Change Approval Sheet DFS-100-CA0 Issue A1 or later approved revisions.3. DFS AFM Supplement DFS-F2000-005 Original Issue or later approved revisions.</td>
<td>1. As per applicability defined on DFS Change Approval Sheet DFS-0100-CA0 Issue A1 or later approved revisions this STC is limited to those Falcon 2000 and Falcon 2000EX previously equipped with Honeywell EGPWS Mk V Soft. 212-212 or later with terrain display in EFIS. - Preliminary Avionics...</td>
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<td>EASA.A.S.03311</td>
<td>AMS AEROSPACE &amp; MECHANICAL ENGINEERINGS LTD</td>
<td>10015012</td>
<td>400A</td>
<td>Activation of Garmin MX-20 Multifunction Display.</td>
<td>Engineering Order EO-M520-003, SAL_049200 8, Rev. 1, Aircraft Flight Manual Supplement AFS-M5200103, SAL_27 092007, Rev. 1. the Instructions for Continued Airworthiness ICA- M5200103, SAL_270 92007</td>
<td>None.</td>
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<td>Date of Issue</td>
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<td>EASA.A.S.03315</td>
<td>SOGERMA SERVICES</td>
<td>SOGERMA SERVICES</td>
<td>STC Holder</td>
<td>AIRBUS A340-313</td>
<td>Virgin Atlantic A340 Cabin Modification.</td>
<td>Modification Approval Sheet, ref. S-00-2020, issue 1 dated 10.03.2008 Flight Manual Supplement, ref. FMS-F4027-00-01, dated 23.05.2008.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/06/2008</td>
<td>Active</td>
<td>1001012</td>
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<td>EASA.A.S.03320</td>
<td>ACI SYSTEMES</td>
<td>ACI SYSTEMES</td>
<td>SOGERMA SERVICES</td>
<td>CESSNA 550</td>
<td>ACI-045 - DFDR Upgrade.</td>
<td>AFM Supplement ACI-13, AFM-44/45 Rev. 1 &amp; Maintenance Manual Supplement ACI-12-MPS-M-045 Rev. 1.</td>
<td>This STC is only applicable to aircraft equipped with Fairchild F800 FDR.</td>
<td>16/06/2008</td>
<td>Orphaned</td>
<td>1001016</td>
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<td>EASA.A.S.03323</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>SOGERMA SERVICES</td>
<td>2, B90, C90</td>
<td>Installation of ACCESS-620 TCAS and EX500 Display, together with necessary antennas.</td>
<td>This STC is only applicable to the S/N 309.</td>
<td>Only airplanes complying with the limitations and conditions listed in Chapter 2.1 “Limitations and Conditions for Installation” of the Installation Manual, Dec No. 126635-01-01, Revision 2, dated 12 Dec-2007, or later issued revision, (as also listed in EASA approved Compliance...</td>
<td>19/12/2007</td>
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<td>EASA.A.S.03324</td>
<td>STC TWENTY ONE LIMITED</td>
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<td>CAA UK RESPONSIBIL ITY</td>
<td>CESSNA 550</td>
<td>Introduction of Dual GNS-430/530</td>
<td>STC Twenty One Ltd Major Change 22.31.34-29-1085 Flight Manual Supplement S21-02A Issue 1</td>
<td>Only airplanes complying with the limitations and conditions listed in Chapter 2.1 “Limitations and Conditions for Installation” of the Installation Manual, Dec No. 126635-01-01, Revision 2, dated 12 Dec-2007, or later issued revision, (as also listed in EASA approved Compliance...</td>
<td>05/06/2008</td>
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<td>EASA.A.S.03326</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BOEING 737-500</td>
<td>Layout of Passenger Accommodation.</td>
<td>Certification Manual BAC/ 737082</td>
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<td>19/05/2008</td>
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<td>EASA.A.S.03327</td>
<td>E.I.S.A.</td>
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<td>CESSNA 525B</td>
<td>S134 - Maximum Takeoff/Landing Weight reduction.</td>
<td>STC Data List (SDL) EISA 00 SDL Rev. 4 Aircraft Flight Manual Supplement EISA 00 AFMS S134 Rev. 4</td>
<td>None</td>
<td>05/02/2008</td>
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<td>10175023</td>
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<td>EASA.A.S.03329</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD</td>
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<td>BOEING 767-300</td>
<td>Layout Passenger Accommodation</td>
<td>Modification BAC/767/008</td>
<td>Aircraft Serial No 9032 Only.</td>
<td>17/04/2008</td>
<td>Active</td>
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<td>EASA.A.S.03330</td>
<td>GULFSTREAM AEROSPACE Ltd.</td>
<td>CAA UK RESPONSIBILITIES</td>
<td></td>
<td>GULFSTREAM GV-SP</td>
<td>Repositioning of Cabin Seat to meet the requirements of JAR25.560(c)(2)(E)</td>
<td>In accordance with Gulfstream Aerospace Limited Change Leaflet for Change No GAE.07/06 Issue 1, dated 24 October 2007 or later approved revisions.</td>
<td>Aircraft Flight Manual Supplement 667-071</td>
<td>29/10/2007</td>
<td>Active</td>
<td>10175004</td>
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<td>EASA.A.S.03338</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
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<td></td>
<td>BEECH 1900C, AIRLINER</td>
<td>Installation of a toilet cargo area</td>
<td>Modification file BCA 103.</td>
<td>Modification Limited to Beech 1900C s/n UC-168</td>
<td>10/01/2008</td>
<td>Active</td>
<td>10175029</td>
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<td>EASA.A.S.03340</td>
<td>VITROGIET S.p.A.</td>
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<td>EASA A.S. 03340</td>
<td>Installation of Rockwell Collins Mode S Transponder System. Notes: This STC supersedes EASA.A.S.03340.</td>
<td>AFM Supplement DRW No. 34-54-1707-800-000 Rev. IR dated 30-07-2007 and subsequent approved revisions. MDL DRW No. 34-54-1707-000-000 Rev. IR dated 25-10-2007 and subsequent approved revisions.</td>
<td>Prior to installation of this modification the installer must determinethat the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is subject to written agreement with the installer.</td>
<td>07/01/2009</td>
<td>Active</td>
<td>10443130</td>
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<td>EASA.A.S.03344</td>
<td>ALTRAN LAB/v AIRCAD</td>
<td></td>
<td></td>
<td>EMB-120</td>
<td>EMB-120 VIP Cabin Modification is a major change designed by AIRCAD that converts half part of the airliner cabin to a VIP cabin with six VIP seats and two tables on pallets and curtain dividing the airline and VIP areas. One of the three emergency exits is blocked.</td>
<td>AIRCAD Certification Bulletin CP 07-001 AIRCAD Service Bulletin ACD-SB 07-001</td>
<td>None.</td>
<td>11/07/2008</td>
<td>Active</td>
<td>10175032</td>
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<td>EASA.A.S.03345</td>
<td>BRITISH AIRWAYS AIRWAYS LIGHTBAU AG</td>
<td></td>
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<td>CAA UK RESPONSIBILITIES</td>
<td>Reconfiguration of Passenger Cabin</td>
<td>British Airways Engineering Master Data List MDL-0054-08, Issue 1, 2005/2008</td>
<td>Aircraft Flight Manual Supplement BCA 102, rev. 0, dated 30-10-2007 or later approved revisions.</td>
<td>13/06/2008</td>
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<td>EASA.A.S.03348</td>
<td>BUCHER LEICHTBAU AG</td>
<td></td>
<td></td>
<td>MD-11</td>
<td>Modification of New Generation Stretcher (NGS) pn. 1120000-201 for installation in a MD-11 on the right hand side of the fuselage, in accordance with Bucher General Arrangement Drawing 1120000-201.</td>
<td>This approval is only valid for the modified installation in the MD-11 on the right hand side of the fuselage. All other installations of the New Generation Stretcher are covered by FOCA STC No. 225-20-068 is the responsibility of the installer to make sure that this installation is certified for use.</td>
<td>06/11/2007</td>
<td>Active</td>
<td>10175035</td>
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<td>EASA.A.S.03351</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
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<td></td>
<td>BEECH 1900</td>
<td>28-230 volts inverter installation on Beech 1900 series according to modification file BCA102, rev. 0, dated 30-10-2007 or later approved revisions.</td>
<td>Airplane flight manual supplement BCA 102, rev. 0, dated 30-10-2007 or later approved revisions.</td>
<td>Airplane flight manual supplement BCA 102, rev. 0, dated 30-10-2007 or later approved revisions must be used when modification is applied.</td>
<td>19/11/2007</td>
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<td>10175036</td>
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<td>Limits/Conditions</td>
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<td>ACC COLUMBIA Jet Service GmbH</td>
<td>LEAQUET 36A Installation of Laser Reflector Dekagon</td>
<td>As per Master Document List MDOC 11710, rev. 6</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>05/06/2008</td>
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<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>CESSNA 525 Installation of new COM-NAV-GPS with LUFTHANSA TECHNIK AG</td>
<td>1. Installation in accordance with RUAG Master Document List Doc-No. 225-B-MDL, Engineering Order Doc-No. 225-B-EO, latest approved revision. 2. To be maintained in accordance with RUAG Airplane Maintenance Manual Supplement 225B-AWMS, or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>08/02/2008</td>
<td>Active</td>
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<td>JET AVIATION AG, BASEL</td>
<td>GULFSTREAM GV-SP Q850 Wireless LAN (WLAN) Activation</td>
<td>Master Drawing List BGLFS23261-001 Rev. AAFMS BGLFS23261-201-01 Rev. 0</td>
<td>The installation of this STC has a prerequisite aircraft configuration. 2. This STC is only applicable on aircraft in association with the EASA STC ref: EASA.A.S.02663.3. The Certification Review Item A-01 has to be considered, addition to the EASA Certification Basis 4. Prior to inst...</td>
<td>11/03/2008</td>
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<td>A300B4-603 / -605R</td>
<td>CESSNA 525 Installation of new COM-NAV-GPS with LUFTHANSA TECHNIK AG</td>
<td>Supplement Type Definition Document STDD05-07/10M Issue 01 dated 20 February 2008 Engineering Orders - EO 158869 Issue 04 - EO 158689 Issue 04 - EO 158381 Issue 03 (or any later revision approved revision of this document).</td>
<td>The installation of this STC will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>04/03/2008</td>
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<td>CESSNA 500, CITATION</td>
<td>BD/04-07 Installation of Mode S Transponder - Elementary Surveillance.</td>
<td>BDI 004-07-01 Master Drawing List BDI 004-07-01-11 AFMS BDI 004-07-18 MEM.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>05/03/2008</td>
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<td>ELECTRONIQUE MAINTENANCE AVIATION</td>
<td>BEECH 200 Installation of new COM-NAV-GPS with BRNAV capacity and coupled to AP.</td>
<td>EMA/07/03/EV-20005ie/ VHF-NAV-GPS, dated November 2007, revision O4FM Supplement EMA/ 07/03/EV-20005ie/VHF-NAV/GPS.</td>
<td>AFM Supplement must be used when the application is applied. Modification limited to sin BB861.</td>
<td>04/06/2008</td>
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<td>A319-133</td>
<td>SABENA TECHNICS BGC Finish Material Change in Private Area on Airbus A319-133 MSN 1468. This modification concerns the change of fabric on part of the existing panels (including seals) in the private office, washroom and conference room in the A319CJ MSN 1468.</td>
<td>Master Data List, ref. No. 7007-03-A-2103-R01, Rev. 01, dated 22.11.07 Certification Plan, ref. No. 7007-03-A-2101-R01, Rev. 01, dated 22.11.07 Compliance Records, ref. No. 7007-03-A-2104- R01, Rev. 01, dated 22.11.07 Instruction for Continued Airworthiness, ref. No. 7007-03-A-2211-R.</td>
<td>This STC is limited to aircraft Airbus A319-133 MSN 1468.</td>
<td>30/11/2007</td>
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<td>EASA.A.S.03363</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>10015047</td>
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<td>BOEING 767-300</td>
<td>Condor B767 Interior Modification</td>
<td>Supplemental Type Definition Document, STCD00-072M, issue/01, Rev 00, July 01/2008</td>
<td>* Prior to installation of this modification the installer must determine that the interoperability between this modification and any other previously installed modifications will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third...</td>
<td>20/08/2008</td>
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<td>EASA.A.S.03364</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) LTD.</td>
<td>10015056</td>
<td>1294058</td>
<td>BOEING 757 - 200</td>
<td>Installation of revised Passenger Accommodation</td>
<td>Bournemouth Aviation (Consultants) Ltd Compliance Document Summary Sheet and Accomplishment Instructions, BAC/757/014, Issue 3, 22/01/2009</td>
<td>Prior to installation of this modification the installer must determine that the interoperability between this modification and any other previously installed modifications will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>23/01/2009</td>
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<td>EASA.A.S.03366</td>
<td>CARGOLUX AIRLINES INTERNATIONALS A.</td>
<td>10015052</td>
<td>1294059</td>
<td>BOEING 747-400F</td>
<td>ICS-220A Iridium sitcom installation on B747-400F airplane in accordance with Master Drawing List CLX-STC-07-01, dated 22 November 2007 or later EASA approved revisions.</td>
<td>Data package containing the following documents and additional supporting data: Engineering Data Report CLX-STC-07-01, dated 22 November 2007 Master Drawing List CLX-STC-07-01 Drawing CLX-STC-07-01 Boeing SSIM and WDM for Salcom and AMU</td>
<td>The Data on Airworthiness Limitations, revisions required of the Maintenance publications etc. arising from the change associated with the STC is provided in DWW CLX-STC-07-01. ICS-220A Iridium Salcom Installation on B747-400F Airplanes. ICA and miscellaneous information.</td>
<td>22/04/2008</td>
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<td>EASA.A.S.03369</td>
<td>GAMA ENGINEERING Ltd.</td>
<td>10015050</td>
<td>1294057</td>
<td>CESSNA 500, CITATION</td>
<td>Upgrade of Mode S Transponder for Enhanced Surveillance</td>
<td>Lees Avionics Major Change HLMO1925Lees Avionics AFM Supplements AFMS925001 Issue 1AFMS925002 Issue 1</td>
<td>05/03/2008</td>
<td>Active</td>
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<td>EASA.A.S.03370</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>10015051</td>
<td>1294060</td>
<td>-</td>
<td>GPWS to EGPWS System Upgrade (Honeywell)</td>
<td>AFM Supplement A0709-071, rev. IR (or later approved revisions)</td>
<td>21/02/2008</td>
<td>Active</td>
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<td>EASA.A.S.03373</td>
<td>SCANDINAVIAN AVIONICS DESIGNS ApS</td>
<td>10015052</td>
<td>1294059</td>
<td>DE HAVILLAND DHC-6-300</td>
<td>Dual Directional Gyro, Dual EHSI/RMI and audio System.</td>
<td>STG Document List 2043963 STC Engineering order and certification program. 2043963-FOAFM 204-9963-AM</td>
<td>As detailed in the AFM 2043963-AM</td>
<td>05/03/2008</td>
<td>Active</td>
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<td>EASA.A.S.03375</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>10015054</td>
<td>1294060</td>
<td>DORNIER DO 228-202K</td>
<td>Increase of Fatigue Life Limit.</td>
<td>STC Engineering Time Limits/ Maintenance Checks Manual for Aircraft S/N 8046. TM- STLCM3-8046-180405 Rev. 1.</td>
<td>This STC is only applicable to Aircraft S/N: 8046.</td>
<td>19/12/2007</td>
<td>Active</td>
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<td>EASA.A.S.03376</td>
<td>BRITISH AIRWAYS PLOMBIRISH AIRWAYS ENGINEERINGOF AIRWORTHINESS</td>
<td>10015055</td>
<td>1294060</td>
<td>BOEING 757-200</td>
<td>B757 GPS upgrade to EGPRS.</td>
<td>British Airways Master Data List NLX-0054-07, issue 00, 12-12-2007.</td>
<td>BA B757 AFM Change Sheet No. 10, Page 1 and 3 (issue 2).</td>
<td>28/03/2008</td>
<td>Active</td>
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<td>EASA.A.S.03378</td>
<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>10015056</td>
<td>1294060</td>
<td>A300F4-608ST</td>
<td>FMS Switching The modification installs the coupling of such FMC to both FMS CDU and EFIS displays, in order to comply with the B-RNAV requirements in case of one FMC source failure. This modification allows dispatching the serviceable FMC signal on the CDU and EFIS displays affected by the loss of...</td>
<td>Fiche de Modification n° ST34M00021, Issue 1 (or any later approved revision) MMEL Supplement: Annex 7 of the Fiche de Modification n° ST34M00021. 1. The MMEL Supplement associated with this STC is only quoted for information. The STC approval EASA.A.S.03378 does not constitute an approval. The MMEL Supplement approval will be granted separately.</td>
<td>12/03/2008</td>
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<td>STC no.</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.03380</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>1</td>
<td>Beech 1900C, Airliner</td>
<td>Installation of a galley in the FW&amp;D right area of the A/C.</td>
<td>1. Installation must be in accordance with BCA 104 Master Data List Rev. 1, dated January 22, 2008 or later Approved Revisions;2. Maintenance must be in accordance with BCA 104 Aircraft Maintenance Manual Rev. 1, dated January 22, 2008 or later Approved Revisions.</td>
<td>Maintenance must be in accordance with BCA 104 Aircraft Maintenance Manual Rev. 1, dated January 22, 2008 or later Approved Revisions.</td>
<td>Must be installed on Beech 1900C (S/N UC-1 and after).</td>
<td>29/01/2008</td>
<td>Active</td>
<td>10015059</td>
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<td>EASA.A.S.03381</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>REV. 1</td>
<td>Beech 1900C, Airliner</td>
<td>Installation of a galley in the FW&amp;D right area of the A/C.</td>
<td>Installation must be in accordance with BCA 106 Master Data List Rev. 1, dated January 22, 2008 or later Approved Revisions.</td>
<td>Maintenance must be in accordance with BCA 106 Aircraft Maintenance Manual Rev. 1, dated January 22, 2008 or later Approved Revisions.</td>
<td>Must be installed on Beech 1900C (S/N UC-1 and after).</td>
<td>29/01/2008</td>
<td>Active</td>
<td>10015060</td>
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<td>EASA.A.S.03383</td>
<td>FEDERATION FRANCAISE DE VOL A VOILE (FFVV)</td>
<td>CS11 75, CS75 VTC</td>
<td>Winglet Retrofitting.</td>
<td>Dossier de modification/supplement aux Manuels de vol/Statement of conformity.</td>
<td>2.2 Supplemental Flight Manual</td>
<td>Maintenance must be in accordance with BCA 106 Aircraft Maintenance Manual Rev. 1, dated January 22, 2008 or later Approved Revisions.</td>
<td>Must be installed on Beech 1900C (S/N UC-1 and after).</td>
<td>01/02/2008</td>
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<td>10015061</td>
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<td>EASA.A.S.03389</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>HS125-700A, HS125-700B</td>
<td>Sandel ST3401 TAWS Installation.</td>
<td>Installed in accordance with 2044235-EO Engineering Order Issue 2 or later approved revision. Operated in accordance with 2044235-AM AFM supplement Issue 1 or later approved revision.</td>
<td>Maintained in accordance with 2044235-AM AFM supplement Issue 1 or later approved revision.</td>
<td>Maintained in accordance with 2044235-AM AFM supplement Issue 1 or later approved revision.</td>
<td>None</td>
<td>30/05/2008</td>
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<td>Foreign/NA A STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.03391</td>
<td>SOCATA</td>
<td>SOCATA TBM 700</td>
<td></td>
<td></td>
<td>Installation of a multi-functions display Garmin GNM 200 connected to already installed equipments (weather radar, COM/NAV/GPS, Traffic Advisory System).</td>
<td>STC Holder</td>
<td></td>
<td>Definition and Installation/EADS SOCATA Modification Data Sheet (MDS) n° STC.34.001.07, edition 1 revision 0, dated 15/01/2008 or later EASA approved revisions. Operations: EADS SOCATA TBM 710 Pilot Operation Handbook (POH) Supplement n° 48A&quot;Garmin GNM 200 Multi-function display&quot; rel. STC 34-001.07-41...</td>
<td>This revision can be installed on TBM 700 s/n to 433 when equipped with EADS SOCATA optional Modifications n° OPT 70-23-019, OPT 70-23-018, OPT 70-34-047 and OPT 70-34-040. The installation can be performed only by EADS SOCATA technical staff according to the MDS here-above referred.</td>
<td>06/02/2008</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.03392</td>
<td>CHRISTEN-ARITECH AG</td>
<td>BEECH C90 KING AIR</td>
<td></td>
<td></td>
<td>GNS 530A, GNS 430A, GAD 42 Installation/Removed equipment: -2 KTR 906 COM 1 &amp; COM 2 -2 KNR 630 NAV 1 &amp; NAV 2 -1 KLN 908GPS with 9K-950 GPS Ammunication -1 KDM 705A DME with KDI 570 DME Indicator New installed equipment: -1 GNS 530AW COM/NAV/ GPS 1 with GAD 42 Adapter Box and MD-41 Annunciator Unit...</td>
<td>STC Holder</td>
<td></td>
<td>- Master Document List, Doc. No. 207.1688 Revision 0 or latest approved Revision- GNS 530AW-AFMS, Doc. No. 207.1688- FS-01-00 Rev. 0 or latest approved Revision- GNS 430AW-AFMS, Doc. No. 207.1688- FS-02-00, Rev. 0 or latest approved Revision...</td>
<td>F-GBLU, SN: LJ-822.</td>
<td>10/06/2008</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.03393</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>CESSNA 550, (CITATION II) CESSNA 550 CITATION II CESSNA 500, CITATION</td>
<td></td>
<td></td>
<td>Installation of Honeywell ART2100/ KMDS50 Weather Radar System.</td>
<td>STC Holder</td>
<td></td>
<td>1. AFM Supplement: 2044352-AM Issue 1.2 Engineering Order: 2044352-EO Issue 2.3 Continued Airworthiness: 2044352-CA Issue 1.</td>
<td>1. This STC is exclusively applicable for aircraft which have the equipment listed in chapter 4 of the Scandinavian Avionics Engineering Order 2044352-EO Issue 2 already installed.2 Prior to installation of this modification the installer must determine that the interrelationship between this...</td>
<td>12/03/2008</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.03394</td>
<td>STC TWENTY ONE LIMITED</td>
<td>AIRBUS A321 AIRBUS A320 AIRBUS A319 AIRBUS A318</td>
<td></td>
<td></td>
<td>Introduction of Wingspeed XLLink Communications and Data Management System</td>
<td>STC Holder</td>
<td></td>
<td>1. STC Twenty One Limited Major Change S21.23-02-10352 Flight Manual Supplement S21.FMS-042 Issue 1</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>29/10/2008</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.03395</td>
<td>CESKA LETECKA SERVISNI A.S.</td>
<td>D</td>
<td>LETECKE ZAVODY L 200 D</td>
<td>LETECKE ZAVODY L 200 A</td>
<td>Installation of new COM, NAV, GPS, XPROR = Encoder, DME, ADfand flight instruments to meet the current technical and operating requirements of IFR category.</td>
<td>Flight manual supplements In Czech language only:AC159-2501-ELTME-406AC159-2502 - Autopilot GMA-347AC159-2503 - NAV/COM/GPS GNS-430AC159-2504 - Radiométrie interconnexion GXT-339AC159-2506 - Gyromagnétiques compas # KCS-55AAC159-2508 - MLD validenten KN-64AC159-2507 - Radiokompas KR-87</td>
<td></td>
<td>04/07/2008</td>
<td>Active</td>
<td>10015075</td>
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<tr>
<td>EASA.A.S.03396</td>
<td>S.E.A.SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td></td>
<td></td>
<td>PIPER PA-28R-200</td>
<td>Modification d'installation Radio / B-RNAV sans couplage P61P: 1 boîte audio KA 134+1 GPS 150L+1 VHF/UKF XZ 155+ IND X: 150 + 1MKR K251 +1 ATC K790(Pose: 1 boîte audio(KFA GMA 340)+1 VHF/UKF/GLD/GPS GNS 430 + IND GI</td>
<td>Fiche d'Approbation de Conception d'une Evolution N°1842-07: Revision 4 du 03/02/2009SUPPLEMENT N°1 AU MANUEL DE VOL - SEAM/SMV19432/07, Revision 2 du 03/02/2009</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by third pers...</td>
<td>02/03/2009</td>
<td>Active</td>
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<tr>
<td>EASA.A.S.03397</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td>MYSTERE FALCON 50</td>
<td></td>
<td>Air Data Sensor Installation for non essential systems</td>
<td>DFS Master Document List DFS-DE-17-FS-6657 Issue A1 or later approved revisions. This STC is limited to all Mystere Falcon 50 aircraft without ARINC 429 air data parameters available. In case the aircraft is RVSM compliant this DFS modification will be applicable only on those aircraft on which Dassault Aviation SB F50-246 have been implemented.</td>
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<td>28/03/2008</td>
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<td>10015077</td>
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<td>EASA.A.S.03406</td>
<td>ACI SYSTEMES</td>
<td></td>
<td>BEECH 300</td>
<td>BEECH 200</td>
<td>Installation of a Garmin 400 Series with B-RNAV capability on Beech 200.</td>
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<td>05/02/2008</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.A.S.03407</td>
<td>AEROSERVIS, S.R.O.</td>
<td>-</td>
<td>L-200 A</td>
<td>Installation of NAV/COM/GPS device GNS 430</td>
<td>Supplement to AFM (only the Czech language) No. STC AE-076A-DLP1 &quot;Doplněk Směrnic pro pílotu letounu L200A&quot;.</td>
<td>Reg. TJ-AAW/SN 486/valid for this S/N only.</td>
<td>25/03/2008</td>
<td>Active</td>
<td>10127075</td>
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<td>EASA.A.S.03408</td>
<td>JET AVIATION AG, BASEL</td>
<td>-</td>
<td>GULFSTREAM III</td>
<td>Indium SATCOM Installation consisting of 1 ICS-200 Indium Transceiver- 108 Indium Antenna- 1 Sigma 7 handsat- 2 Coldless Hardset</td>
<td>Master Data List. Doc. No. BTAIA/202-001, Rev. A or latest approved revision.</td>
<td>Reg. TJA-AW/SN 486/valid for this S/N only.</td>
<td>05/03/2008</td>
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<td>EASA.A.S.03410</td>
<td>FR AVIATION Ltd./DB COBHAM AVIATION SERVICES UK</td>
<td>-</td>
<td>BEECH B300</td>
<td>Introduction of Equipment Associated with Operational Requirements. The introduction of supplementary radio navigation aids to support avionics calibration systemand items of equipment for EU-OPS operation.</td>
<td>Installation must be in accordance with Master Document List included in the Design ChangeStatement DC-0127 Iss.1 dated 28th November 2006 or later.</td>
<td>28/11/2008</td>
<td>Active</td>
<td>10104312</td>
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<td>EASA.A.S.03411</td>
<td>FR AVIATION Ltd./DB COBHAM AVIATION SERVICES UK</td>
<td>REV 1</td>
<td>BEECH B300</td>
<td>ADFIS Equipment Installation. Installation of equipment transport operation as an artificialcalendar aircraft.</td>
<td>Installation must be in accordance with Master Document List included in the Design ChangeStatement DC-0128 Iss.1. Dated 28th November 2007 or later.</td>
<td>09/12/2008</td>
<td>Active</td>
<td>10104312</td>
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<td>EASA.A.S.03412</td>
<td>SOGERMA SERVICES</td>
<td>-</td>
<td>AIRBUS A320 SERIES</td>
<td>Installation of passenger entertainment cabin audio system on seats.</td>
<td>Service Bulletin SOG-DO-23-33-01</td>
<td>04/01/2008</td>
<td>Active</td>
<td>10105078</td>
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<td>EASA.A.S.03414</td>
<td>SOGERMA SERVICES</td>
<td>-</td>
<td>AIRBUS A300 MODEL B2-1A</td>
<td>Installation of a GPS/HS adapter for this STC superseeded DGAC France STC 180-SF-0174.</td>
<td>SOG-34-500 issue C</td>
<td>04/01/2008</td>
<td>Active</td>
<td>10105080</td>
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<td>EASA.A.S.03416</td>
<td>SOGERMA SERVICES</td>
<td>-</td>
<td>AIRBUS A300-600</td>
<td>Installation of an ELT 96 emergency locator transmitter: Note: This STC superseeded DGAC France STC F-C180-MJ-0864 and EASA 2004-4596.</td>
<td>Modification Approval Sheet SOG-25-070 issue C, dated 31 March 2004</td>
<td>04/01/2008</td>
<td>Active</td>
<td>10105082</td>
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<td>EASA.A.S.03417</td>
<td>SOGERMA SERVICES</td>
<td>-</td>
<td>AIRBUS A320 SERIES</td>
<td>Cabin conversion from 181 Y/C to 16 B/C &amp; 169 Y/C: Note: This STC superseeded DGAC France STC 180-SF-0229 and EASA 2004-9188</td>
<td>Modification Approval Sheet (MAS) SOG-25-795 issue 2, dated 12 March 2004</td>
<td>04/01/2008</td>
<td>Active</td>
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<td>EASA.A.S.03418</td>
<td>SOGERMA SERVICES</td>
<td>-</td>
<td>AIRBUS A320 SERIES</td>
<td>Installation of BENNET seat container: Note: This STC superseeded DGAC France STC 180-SF-0229 and EASA 2004-2036.</td>
<td>Modification Approval Sheet (MAS) SOG-25-795 issue 2, dated 10 February 2004</td>
<td>04/01/2008</td>
<td>Active</td>
<td>10105084</td>
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<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.03420</td>
<td>SOGERMA SERVICES</td>
<td>1</td>
<td>DGA/C-428 06-002</td>
<td>AIRBUS A300 MODEL B2</td>
<td>Cargo door modification for use of SPC</td>
<td>Modification Approval Sheet (MAS) SOQ-52.340 issue E</td>
<td>1. Valid only for MSN 003.</td>
<td>04/01/2008</td>
<td>Active</td>
<td>10101050</td>
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<td>EASA.A.S.03421</td>
<td>SOGERMA SERVICES</td>
<td>1</td>
<td>DGA/C-428 06-002</td>
<td>AIRBUS A300 MODEL B2</td>
<td>Installation of a ground proximity warning system/threat. This STC supersedes EASA STC 2004-2393</td>
<td>MAS N SOQ-34.576 Issue 3A/PM5 MTG/208/0103</td>
<td>1. Only valid for MSN 003.</td>
<td>04/01/2008</td>
<td>Active</td>
<td>10101008</td>
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<td>EASA.A.S.03423</td>
<td>JET AVIATION VIENNA GMBH</td>
<td>1</td>
<td>DGA/C-428 06-002</td>
<td>525</td>
<td>Rockwell Collins MDF-124F (V2) Direction Finder System Installation</td>
<td>Certification Report CS25-A RS34-004, Rev. 1, 15.02.2008</td>
<td>Referred to AFMS limitation</td>
<td>27/02/2008</td>
<td>Active</td>
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<td>EASA.A.S.03424</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>1</td>
<td>DGA/C-428 06-002</td>
<td>FALCON 10</td>
<td>Dual TDR-94D Transponder Upgrade to provide limited Enhanced Surveillance Transponder Mode S capability (Non EHS Capable)</td>
<td>Previous required aircraft configuration + Aircraft equipped with Dual TDR-94D ( &lt;-007-008) and installation compliant with Elementary Mode S requirements + Mode S Transponder Antenna Diversity + Collins TCAS II System + Dual ADC’s installation with or ADC 1 Collins ADC 80 with ARINC 457 in...</td>
<td>22/02/2006</td>
<td>Active</td>
<td>10101059</td>
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<td>EASA.A.S.03428</td>
<td>LUFTHANS TECHNIK AG</td>
<td>1</td>
<td>DGA/C-428 06-002</td>
<td>A3118</td>
<td>A318 – Ballast Weight Installation</td>
<td>- Supplemental Type Definition Document, ref. Doc. Doc. No. STDD03-05/02M issue 01, dated 11th August 2008 or later approved revisions - Compliance Record List, ref. Doc. No.CRLS3-08/02M, issue 01, dated 11th August 2006 - Applicant’s Declaration of Compliance No. 2147 – A318 Ballast Weight...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>12/08/2008</td>
<td>Active</td>
<td>10101051</td>
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<td>EASA.A.S.03429</td>
<td>BUCHER LEICHTBAU AG</td>
<td>1</td>
<td>DGA/C-428 06-002</td>
<td>PILATUS PC-12</td>
<td>Heavy Duty Floor Panel</td>
<td>(AFMS 01247/40) Rev. N/C or earlier CMG 16-08-09 Rev. N/C or later</td>
<td>19/05/2008</td>
<td>Active</td>
<td>10101052</td>
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<td>EASA.A.S.03439</td>
<td>NSE</td>
<td>1</td>
<td>DGA/C-428 06-002</td>
<td>SA 226-T</td>
<td>Modification of radio electrical Installation to VHF 2, ICR/1, ICR2/1, ICR3/1, ICR4/1, ICR5/1, ICR6/1 moves/2/3</td>
<td>Airplane Flight Manual Supplements - 04/2007 - AFMS TANS Ed. 1 Rev. 0, 02 Feb 2008 - 04/2007-AFMS GN2 430 WR/3001 Ed. 1 Rev. 0 Feb 2009. The approval is limited to MSN T-276 F-GLGCP use is forbidden for SIDS/STAR ApproachesGPS/Auto Pilot coupling is forbidden below 2000 ft.</td>
<td>05/03/2008</td>
<td>Active</td>
<td>10101050</td>
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<td>EASA.A.S.03441</td>
<td>ACI SYSTEMES</td>
<td>1</td>
<td>DGA/C-428 06-002</td>
<td>BEECH 90</td>
<td>Installation of Garmin 530 Sxavis GPS with B-RNAV capability on Beech 90.</td>
<td>Supplemental Airplane Flight Manual ACI-06-AFM-M-070, revision 0, dated 15 January 2008 must be on board when the modification is installed.</td>
<td>04/04/2008</td>
<td>Orphaned</td>
<td>10101008</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.A.S.03443</td>
<td>PHOENIX AEROSPACE LIMITED</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>PIPER PA-31</td>
<td>Installation of new Avionics</td>
<td>Phoenix Aviation Design Services Limited Design Change PADS/CP1110 Part B Flight Manual Supplements PADS/CP1</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>30/06/2008</td>
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<td>10015103</td>
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<td>EASA.A.S.03444</td>
<td>PHOENIX AEROSPACE LIMITED</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>PIPER PA-39</td>
<td>Validation of Foreign Installed Avionics</td>
<td>Phoenix Aviation Design Services Ltd Major Change PADS/CP1111 Flight Manuals Supplements PADS/CP1</td>
<td>05/06/2008</td>
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<td>10015101</td>
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<td>EASA.A.S.03445</td>
<td>AIM AVIATION (JECCO) LIMITED</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>A340-313X</td>
<td>Premium Cabins Upgrade/Appearance, comfort and maintenance improvement</td>
<td>AIM Major Change, ref. Doc. No. MM385/07, Issue C, dated 1st July 2008 AIM Certification Plan and Compliance Check List, ref. Doc. Appendix to Major Change MM385/07, Issue C, dated 1st July 2008. Declaration of Compliance - page 59 of the Appendix to Major Change MM385/07, dated 4th Aug...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>05/08/2008</td>
<td>Active</td>
<td>10015102</td>
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<td>EASA.A.S.03447</td>
<td>AVIONITEC Ltd.</td>
<td></td>
<td></td>
<td>F90</td>
<td>Avionics replacing: Garmin GTX330D XPD and Weather Radar System RDR 2000.</td>
<td>Installation must be in accordance with Avionitec Master Document List No. 073839-011 Rev. 0 dated 19. Mar. 2008 or later Approved Revisions. Operation must be in accordance with Avionitec Airplane Flight Manual Supplement No. 073839-011 dated 17. Feb. 2008 and No. 073839-012 dated 17. Feb. 2008 or la...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>23/06/2008</td>
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<td>EASA.A.S.03448</td>
<td>S.E.A.M. SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td></td>
<td>SOCATA TB 20</td>
<td>Installation of a GNS 430 W for IFR B-RNAV “on routes” operations, without coupling to automatic pilot.</td>
<td>Flight Manual Supplement No. SEAM/SM/1983/08 revision 0 dated 15/09/2009.</td>
<td>1. This installation is limited to TB 20 s/n 791.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>17/04/2009</td>
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<td>STC Level</td>
<td>Foreign/NAAT STC ref.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.03450</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>A300 F4-600R</td>
<td></td>
<td>Structure modification for GPS antenna installation. This STC defines the structural modification for the installation of a dual GPS antenna, in place of the existing Gatelink antenna</td>
<td>Supplement Type Definition Document STD035-08/09M Issue 01, Revision 1, dated 29 April 2008. Engineering Bulletin: A300-EB35-0292 Issue 01 (or any later approved revision of these documents). Instruction for Continued Airworthiness, ref: ICA-53-0292 Issue 1</td>
<td></td>
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<td>08/05/2008</td>
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<td>10015107</td>
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<td>EASA.A.S.03451</td>
<td>DASSAULT FALCON SERVICE</td>
<td>REV. 1</td>
<td>MYSTERE FALCON 50</td>
<td>IFE / Galley Electrical Load Shedding.</td>
<td>1. For Falcon 50: DFS Master Document List DFS-DE-17-F5-5653 Issue A1 or later approved revisions; Aircraft Flight Manual Supplement F50-058 Revision Original or later approved revisions. 2. For Falcon SXEC: DFS Master Document List DFS-DE-17-F5-5653 Issue B1 or later approved revisions.</td>
<td></td>
<td>1. This STC is applicable to: All Mystere Falcon 50 aircraft and - Falcon SXEC aircraft variant up to serial number 333 included, on which Dassault Aviation Mod. M2849 &quot;IFE Load Shedding: electrical capability&quot; has not been embodied. 2. This STC approval does not cover any pos...</td>
<td>26/06/2008</td>
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<td>EASA.A.S.03453</td>
<td>DASSAULT FALCON SERVICE</td>
<td>REV. 1</td>
<td>MYSTERE FALCON 50</td>
<td>Installation of AC Electrical Generation / Distribution compliant with JAA TGM/25/10 requisites</td>
<td>DFS Master Document List DFS-DE-17-F5-5651 Issue A1 or later approved revisions.</td>
<td></td>
<td></td>
<td>28/03/2008</td>
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<td>EASA.A.S.03457</td>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td>B747-400</td>
<td>Interior Modifications</td>
<td>- MVL Doc. No. BAHM/R004-001, Rev. BAFMS Doc. No. BAHM/R004-201-01. Rev. IROMM Doc. No. VB-06, Rev. B</td>
<td>1. STC limited to Aircraft SIN 26903.2 For other Limitation refer to the AFMS BAHM/R004-201-0.3 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modifications is conducive to no adverse effect u...</td>
<td>02/06/2009</td>
<td>Active</td>
<td>10043134</td>
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<td>EASA.A.S.03459</td>
<td>RECTIAO AVIATION</td>
<td></td>
<td>PIPER PA-28R-201T</td>
<td>Replacement of : King KX155, KI209 and Garmin GPS 155 by Garmin GNS430 coupled to HSI and autopilot. Installation of a new RMI KI 229. Garmin GNX 430 with KD 41 coupled to HSI KI525A and autopilot.</td>
<td>FACE Rando Aviation 2008/03 AFMS EASA approved Flight Manual Supplement for aircraft equipped with GPS Garmin GNX 430 using GNX CNS 430 in VFR modes coupled to HSI and autopilot ed. July 2008</td>
<td></td>
<td>Applicable to SN 7803232 F-0HYY GPS use is limited to Day VFR flights in sight of ground or water.</td>
<td>22/09/2008</td>
<td>Active</td>
<td>10015112</td>
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<td>EASA STC no.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>EASA.A.S.03460</td>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td></td>
<td></td>
<td>MYSTERE FALCON 20- F5</td>
<td>Dual TDR-94D Transponder Installation Upgrade to provide both Elementary and Enhanced Surveillance Transponder Mode S Capabilities.</td>
<td>See Scandinavian Avionics• 2045142- STC (Certification Documents) issue 2; dated 04 Jun 08• 2045142-EO Engineering Order issue 2; dated 11 Feb 08• 2045142-AM-A AFM Supplement for Mode S Transponder Elementary Surveillance Capability; issue 1 dated 04 Feb 08• 2045142-AM-B AFM Supplement.</td>
<td>Previous required aircraft configuration • Aircraft equipped with Dual TDR-94D installation with Mode S Diversity Antenna Installation and Radio Tuning Unit RTU-422D• ADA’s with ARINC 429 Interface.</td>
<td>06/02/2008</td>
<td>Active</td>
<td>1001513</td>
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<td>EASA.A.S.03462</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING/CO RPORATION AG (IAEC)</td>
<td></td>
<td></td>
<td>LEARJET 60</td>
<td>UNS–1B or 1C upgrade to UNS 1C+</td>
<td>Master Data List Doc. No.: 734-049 AFM Supplement Doc. No. 734-071</td>
<td></td>
<td>06/06/2008</td>
<td>Active</td>
<td>1001514</td>
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<td>EASA.A.S.03466</td>
<td>BOURNEMOUTH AVIATION (CONSULTANTS) Ltd</td>
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<td></td>
<td>BOMBARDIER CL-600-2B16 (604)</td>
<td>Installation of Electronic Flight Bag</td>
<td>BAC Modification No. BAC/ CL-600 Issue 2 BAC FMS/ 53645 Issue2</td>
<td>Restricted to ASN 5383</td>
<td>06/06/2008</td>
<td>Active</td>
<td>1001517</td>
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<td>EASA.A.S.03467</td>
<td>HAWKER BEECHCRAFT Ltd</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td>HAWKER 800 XP</td>
<td>To upgrade Dual Honeywell NZ-2000 FMS Computers from 5.2 to 6.0 software and provide P-RNAV approval.</td>
<td>Hawker Beechcraft Limited Change No. SCO314A – To introduce Dual Honeywell NZ-2000 FMS Computers (with V6.0 software) in lieu of NZ-2000 (with V5.2 software) Operation to be in accordance with Hawker Beechcraft Limited Flight Manual Supplement No. 4 to Flight Manual ref H.S. 1.22, and ASC-2-GEN- AV3313 “P”.</td>
<td>None.</td>
<td>06/10/2008</td>
<td>Active</td>
<td>1001518</td>
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<td>EASA.A.S.03468</td>
<td>HAWKER BEECHCRAFT Ltd</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td>HAWKER 1000</td>
<td>To upgrade Dual Honeywell NZ-2000 FMS Computers in lieu of NZ-900 FMS Computers and add a second GPS ONSU. The existing Data Loader is replaced with Model DL-950.</td>
<td>Certification Plan ASC-C-009-AW3173, Instructions for continued airworthiness ASC-R-009- AV3152, AFM Supplement for Hawker 1000B (HS1.18) Supplement 3 issue 1 Modification ref SCO315A 1. As per AFM Supplement for Hawker 1000B (HS1.18) Supplement 3 issue 1.3. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>25/09/2008</td>
<td>Active</td>
<td>1001519</td>
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<td>EASA.A.S.03474</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td>REV. 1</td>
<td></td>
<td>FALCON 20</td>
<td>Dual Universal UNS–1L FMS Installation.</td>
<td>Master Document List 236A-MDL Revision Original Engineering/Order 236A-F20-34-60.00EO Revision Original Airplane Flight Manual Supplement 236A-F20-34-60.02AFMS Revision NO Maintenance Manual Supplement 236A- F20-34-60.02dFMS Revision NO Maintenance Manual Supplement 236A- F20-34-60.02MFS Revision NO/CA later approved revisions of these documents.</td>
<td>This STC is limited to Mystere- Falcon 20-CS aircraft Serial Number 141. See: Airplane Flight Manual Supplement 236A-F20-34-60.02AFMS Revision NO/CA later approved revisions.</td>
<td>01/04/2008</td>
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<td>Limitations/Conditions</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>BEECH B300C BEECH B300 BEECH 300LW BEECH 300</td>
<td>Installation of an L3 F-1000 Digital Flight Data Recorder System</td>
<td>EASA approved Engineering Order, Doc. No. 2045502-E0, Issue 2, dated 24-Apr-2008, or later approved issue. EASA approved Airplane Flight Manual Supplement, Doc. No. 2045502-AM, Issue 2, dated 14-Apr-2008, or later approved issue.</td>
<td>This change is limited to the configuration of aircraft equipment type as listed under Item 4. of EASA approved Engineering Order, Doc. No. 2045502-E0, Issue 2, dated 24-Apr-2008 or later approved issue.</td>
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<td>05/05/2008</td>
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<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>BEECH B300C BEECH B300 BEECH 8200 BEECH 300 BEECH 300LW BEECH 300 BEECH 200C BEECH 200</td>
<td>Installation of a Universal CVR-30B or CVR-120 Cockpit Voice Recorder</td>
<td>EASA approved Engineering Order, Doc. No. 2045503-E0, Issue 2, dated 23-Apr-2008, or later approved issue. EASA approved Airplane Flight Manual Supplement, Doc. No. 2045503-AM, Issue 2, dated 14-Apr-2008, or later approved issue.</td>
<td>This change is limited to the configuration of aircraft equipment type as listed under Item 4. of EASA approved Engineering Order, Doc. No. 2045503-E0, Issue 2, dated 23-Apr-2008 or later approved issue.</td>
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<td>05/05/2008</td>
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<td>EASA A.03486</td>
<td>HAWKER BEECHCRAFT Ltd.</td>
<td></td>
<td>CAA UK RESPONSIBILITY</td>
<td>HAWKER HS 125-700A To upgrade Dual UASG UNS-1k FM to Dual UASG UNS-1k+ in preparation for P- RNavi compliance.</td>
<td>EASA approved Engineering Order, Doc. No. 2045503-E0, Issue 2, dated 23-Apr-2008, or later approved issue. EASA approved Airplane Flight Manual Supplement, Doc. No. 2045503-AM, Issue 2, dated 14-Apr-2008, or later approved issue.</td>
<td>This change is limited to the configuration of aircraft equipment type as listed under Item 4. of EASA approved Engineering Order, Doc. No. 2045503-E0, Issue 2, dated 23-Apr-2008 or later approved issue.</td>
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<td>EASA A.03488</td>
<td>S.E.A.M SERVICE ELECTRONICAVATION MARINES A.R.L</td>
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<td>BEECH 58</td>
<td>STC 1940-08- Avionics modification for BRNAV navigation and autopilot coupling.</td>
<td>AFM Supplement SEAM AVIATION MARINES A.R.L.</td>
<td></td>
<td>Limitation to s/n TH 1401</td>
<td>06/05/2008</td>
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<td>EASA A.03490</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td>CESSNA 650 CITATION VII</td>
<td>UPGRADE of existing FMS Universal UNS-1c to UNS-1c+ to comply with P-RNAV requirements.</td>
<td>2045492-EO Engineering Order Issue 2 or later approved revision. 2045492-AM AFM Supplement Issue 1 or later approved revision. 2045492-CA Instructions for Continued Airworthiness Issue 1 or later approved revision.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third part...</td>
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<td>25/08/2008</td>
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<td>Foreign/NAAR STC ref.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.A.S.03491</td>
<td>NSE</td>
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<td></td>
<td>SA 226-T</td>
<td>Modification of radio electrical installation: replacement radar indicator by MFD Garmin GND202 and/or Concord Batteries.</td>
<td>Airplane Flight Manual Supplement 06/2008- AFMS GMK 200 Limitation Ed. 1 Rev. 0 Mars 2008.</td>
<td>STC TRANSAIR No. S24 Revision 6, dated 05-01-2008.</td>
<td>1. The approval is limited to neg. T-72F F-GLLO.2. Use of GMK 200 limited to radar weather display function/3. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will int...</td>
<td>03/04/2008</td>
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<td>EASA.A.S.03495</td>
<td>SWS CERTIFICATION SERVICES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BAE 146</td>
<td>STG Aerospace Wireless Emergency Primary Power System (NEPPS) Installation.</td>
<td>- Certification Plan 3011-CEP-01.- Compliance Report 3011-COMA01</td>
<td>STC TRANSAIR No. S24 Revision 6, dated 05-01-2008.</td>
<td>1. The supplemental AFM No. TFS24PA must be used when the modification is installed.2. Limited to S/N UC-168, F-GVC and UC-153, F-GHPM.</td>
<td>08/12/2008</td>
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<td>10015138</td>
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<td>EASA.A.S.03496</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>DOAC AHC NO. 751</td>
<td>BEECH 1900C, ARRUNER</td>
<td>Installation of a King/ Bendix KLN-90B GPS with B-RNAV capacity. Note: This STC revises DOAC approval AHC No. 751.</td>
<td>STC TRANSAIR No. S24 Revision 6, dated 05-01-2008.</td>
<td>1. The supplemental AFM No. TFS24PA must be used when the modification is installed.2. Limited to S/N UC-168, F-GVC and UC-153, F-GHPM.</td>
<td>28/02/2008</td>
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<td>EASA.A.S.03502</td>
<td>LUFTHANSA TECHNIX AG</td>
<td>REV 1 LBA STC LBA. 2162.TA0270</td>
<td>BOMBARDE</td>
<td>Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via LBA STC LBA.2162.TA0270 due to the potential for combustion of contaminants in the dryer unit. A new dryer unit has been developed which removes this...</td>
<td>STD201-028/01M, Issue 04/STD201-07/06M, Issue 01, CL-600-EB21-015AD 2006-0140-E</td>
<td>See related STC LBA. 2162.TA0270</td>
<td>05/06/2008</td>
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<td>EASA.A.S.03503</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CO INCORPORATION AG (AIEC)</td>
<td>1329-25</td>
<td>Cessna 172 &amp; Cessna 172 - ATTI System</td>
<td>Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via LBA STC LBA.2162.TA0270 due to the potential for combustion of contaminants in the dryer unit. A new dryer unit has been developed which removes this...</td>
<td>STD201-028/01M, Issue 04/STD201-07/06M, Issue 01, B737-EB21-015AD 2006-0140-E</td>
<td>See related STC LBA. 2162.TA0270</td>
<td>14/07/2008</td>
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<td>EASA.A.S.03504</td>
<td>LUFTHANSA TECHNIX AG</td>
<td>REV 1 LBA STC LBA. 2162.0007</td>
<td>BOEING 737-700</td>
<td>Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via LBA STC LBA.2162.TA0270 due to the potential for combustion of contaminants in the dryer unit. A new dryer unit has been developed which removes this...</td>
<td>STD201-028/01M, Issue 04/STD201-07/06M, Issue 01, B737-EB21-015AD 2006-0140-E</td>
<td>See related STC LBA. 2162.TA0307</td>
<td>05/06/2008</td>
<td>Active</td>
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<td>EASA.A.S.03505</td>
<td>LUFTHANSA TECHNIX AG</td>
<td>REV 1 LBA STC LBA. 2162.TA0636</td>
<td>BOEING 737-800</td>
<td>Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via LBA STC LBA.2162.TA0636 due to the potential for combustion of contaminants in the dryer unit. A new dryer unit has been developed which removes this...</td>
<td>STD201-028/01M, Issue 04/STD201-07/06M, Issue 01, B737-EB21-015AD 2006-0140-E</td>
<td>See related STC LBA. 2162.TA0636</td>
<td>05/06/2008</td>
<td>Active</td>
<td>10015145</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>STC Holder</th>
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<tr>
<td>STC Holder 10015148</td>
<td>AIRBUS A330-200</td>
<td>Replacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via LBA STC LBA:21E2.TA0609 due to the potential for combustion of contaminants in the dryer unit. A new dryer unit has been developed which removes this...</td>
<td>STDC001/A 05STDD021-07/06M, Issue 01, A330-EB21-01AAD 2006-0140-E</td>
<td>See related STC LBA: 21E2.TA0609</td>
<td>05/08/2008</td>
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<td>E.I.S. AIRCRAFT GmbH</td>
<td>BOEING 737-700</td>
<td>Cabin reconfiguration.</td>
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<td>N/A</td>
<td>15/04/2008</td>
<td>Active</td>
<td>1015148</td>
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<tr>
<td>AERODIENST GMB &amp; CO. KG</td>
<td>3SA</td>
<td>Flight Crew Intercom System (NAT)</td>
<td></td>
<td>Limited to single GNS 480 installation per cockpit only. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>17/10/2008</td>
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<td>1015151</td>
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<td>EASA A.S.03512</td>
<td>PILATUS PC-1247 PILATUS PC-1245 PILATUS PC-12</td>
<td>Replacement of existing KX 165 COM / NAV, IKN 90B GPS, and GTX 330D Mode S Transponder installation by GNS400 COM / NAV GPS and GTX 330D Mode S Transponder installation.</td>
<td>EASA approved Engineering Data, Doc. No. 2007-23, Revision B, dated 25-Jul-2008, or later approved IssueEASA approved Airplane Flight Manual Supplement, Doc. No. 2007-23-A, Revision A, dated 16-Jan-2008, or later approved Issue/Maintenance Manual Supplement, Doc. No. 2007-23-A, Revision B, dated 02...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the...</td>
<td>14/10/2008</td>
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<td>1015152</td>
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<tr>
<td>EASA A.S.03513</td>
<td>PILATUS PC-1247 PILATUS PC-1245 PILATUS PC-12</td>
<td>Installation of AIRCELL SATCOM ST3100</td>
<td>EASA approved Engineering Data, Doc. No. 2007-24, Revision C, dated 01-Oct-2008, or later approved Issue.2. EASA approved Airplane Flight Manual Supplement, Doc. No. 2007-24-A, Revision A, dated 11-Jan-2008, or later approved Issue.3. Maintenance Manual Supplement, Doc. No. 2007-24-A, Revision B...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the...</td>
<td>14/08/2008</td>
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<td>GAMA ENGINEERING Ltd.</td>
<td>HAWKER HS 125-700 SERIES</td>
<td>Dual FMS Upgrade.</td>
<td>Master Document List MDL961001 Issue 1. Modification Statement HLMC961001, AFM Change SH AFM961001 Issue 1, AFM Supplement AFM961002 Issue 1, Instructions for Continued Airworthiness ICA961001 Issue 1</td>
<td>For all aerial numbers.</td>
<td>11/06/2006</td>
<td>Active</td>
<td>1015154</td>
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<td>SCHIERDEL ENGINEERING Ltd.</td>
<td>SCHENPP NIMBUS 4M SCHENPP JANUS CM DQ 80A DQ 600M DQ 500 M DQ 400</td>
<td>Alternate fuel lines for Rotax engine powered glider.</td>
<td>Schedel Engineering, Technical Note TNO- Alternate Fuel Lines for Rotax powered Glider, Glider number08-Alternate Fuel Lines-001, Revision 1</td>
<td>For all aerial numbers.</td>
<td>11/06/2006</td>
<td>Active</td>
<td>1015154</td>
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<tr>
<td>AVONICS INTEGRATION &amp; ENGINEERING/RIPORATION AG (AEC)</td>
<td>LEARJET 60</td>
<td>Installation of Iridium SAT/COM Antenna (Sensor Systems) As noted in Master Data List 745-049, rev. IR (or later approved revisions) Installation of the Iridium SAT/COM antenna and the antennae outlet as covered by this STC does not constitute an approval of the utility system installation or operation.</td>
<td>EASA STC no. HP20070017-61</td>
<td></td>
<td>22/04/2008</td>
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<td>EASA STC no.</td>
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<td>Foreign/NAA STC ref.</td>
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<td>Description</td>
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<td>Associated Technical Documentation</td>
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<td>EASA.A.S.03523</td>
<td>UNI AIR ENTREPRISE SASUN AIR ENTREPRISE</td>
<td></td>
<td></td>
<td>MYSTERE FALCON 20</td>
<td>Installation of a forward partition door between entry area and cabin</td>
<td>Modification on an Un Air Enterprise Master Date List MIL-010-08 Revision 1, dated 14th May 08 or later approved revisions, Un Air Enterprise Master Flight Manual Supplement, dated 13th May 2008 or later approved revisions.</td>
<td>Design change applicable to Mystere Falcon 20 aircraft serial number 011.</td>
</tr>
<tr>
<td>EASA.A.S.03524</td>
<td>AEROSPACE DESIGN AND ENGINEERING CONSULTANTS Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td></td>
<td>737-700</td>
<td>Introduction of seat-track mounted 40 Inch Closet-Boeing 737-700.</td>
<td>Aerospace Design &amp; Engineering Consultants Ltd. Modification Summary Document MOD-0352 Rev. Aircraft Manual Supplement – MSN28630 Aircraft Manual Supplement – MSN28630 Component Maintenance Manual (CMM) 25-30-05</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.A.S.03526</td>
<td>LUFTHANSA TECHNIK AG</td>
<td></td>
<td></td>
<td>A318-112</td>
<td>A318 ELITE - Project Comlux 2; VIP Cabin completion</td>
<td>Supplemental Type Definition Document, ref. Doc. No. STDD00-08/01M, issue 01, dated 10 December 2008, or later approved revision, Aircraft Flight Manual Supplement A-00-08/772-AFMS 1, dated 13 November 2008, or later approved revision Instructions for Continued Airworthiness, ref. Doc. No. A-00...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.A.S.03527</td>
<td>FRANCE AVIATION</td>
<td></td>
<td></td>
<td>BEECH A36</td>
<td>FA 06/2008 Rev. 2 - Avionic change with B-RNAV/STD/STANDARD approach and conventional non-precision approaches privileges with AP coupling.</td>
<td>None</td>
<td>None</td>
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<td>EASA.A.S.03528</td>
<td>AVIONIK STRAUßING ENTWICKLUNGS GmbH</td>
<td></td>
<td></td>
<td>C. SA227 CC, SA227 DC B. SA226-TC, SA227 AC, SA227 BC, A. SA249-AT, SA26-T, SA28 T1B A. SA237-AT, SA237 TT A. SA226-AT, SA228-T</td>
<td>Installation of honeywell Mark VI EGPWS - Optional installation of Com Hav C401-220 GPS Antenna.</td>
<td>Installation must be in accordance with Avionik Straubing Master Document List Doc. No. ASR 2007-144-MDL-01-00 Amendment 0.0 dated 28.12.07 or later Approved Revisions. Operation must be in accordance with Avionik Straubing Flight Manual Supplement No. ASR 2007-144-FMS-05-00 Amendment 0.0 dated 15.12.07 or later Approved Revisions.</td>
<td>06/05/2008</td>
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<tr>
<td>EASA.A.S.03529</td>
<td>TENENCIA Ltd.</td>
<td>REV. 1</td>
<td>CAA UK RESPONSIBILITY</td>
<td>AVRO 146-RJ70</td>
<td>VIP Installation</td>
<td>1. Modification TEN 549.2 - BAE 146 Series VIP Installation Issue 1 dated 28 October 2008.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third pe...</td>
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<tr>
<td>EASA STC No.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
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<td>EASA.A.S.03530</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td></td>
<td>CESSNA 525</td>
<td>CESSNA 525</td>
<td>Upgrade of existing Dual Honeywell MST 67A Mode-S XPDYR Installation to EHS Capabilities</td>
<td>EASA approved Engineering Data, Doc. No. 2007-11, Revision B, dated 01-Jul-2008, or later approved Issue/EASA approved Airplane Flight Manual Supplement, Doc. No. 2007-17-8, Revision A, dated 18-Jan-2008, or later approved Issue/Maintenance Manual Supplement, Doc. No. 2007-17-10, Revision A, dated 15-</td>
<td>- Master Document List, Doc. No. 073723-017, Rev. 1/AFM, Doc. No. 073723-011, Rev. 1</td>
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<td>EASA.A.S.03534</td>
<td>AVIONITEC Ltd.</td>
<td></td>
<td>PA-46-350P</td>
<td></td>
<td>Avionics upgrade to GNS 430, Removed Equipment - KX-165 NAV/COM 2-KO-206 NAV 2 Indicator/Installed Equipment, GNS 430 NAV/COM/OPS 2- GI-105A NAV/GPS Indicator 2-GA-58 GPS 2 Antenna</td>
<td>- Master Document List, Doc. No. 073723-017, Rev. 1/AFM, Doc. No. 073723-011, Rev. 1</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. Reg.: HB-PPH, S/N: 4636045</td>
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<tr>
<td>EASA.A.S.03536</td>
<td>AERODATA AG</td>
<td>REV. 1</td>
<td>550</td>
<td></td>
<td>Additional Cockpit Information Display</td>
<td>Master Document List MOL-AV137 Revision 2-Airplane Flight Manual Supplement AFM-095 Revision 1-Airplane Maintenance Manual Supplement AMM-952 Revision 1</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<tr>
<td>EASA.A.S.03537</td>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td>MYSTERE FALCON 20</td>
<td></td>
<td>Installation of a Honeywell MK VIII Enhanced Ground Proximity Warning System (EGPWS).</td>
<td>- Master Data List BDMET063-001, Revision B, dated 15th June 2008 - Aircraft Flight Manual (AFM) Supplement BDMET063-201-25, Revision B, dated 15th June 2008- Instructions for Continued Airworthiness BDMET063-150-25, Revision IR, dated 01st May 2009- Engineering Order BDMET063-344600, Rev. 1</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. This STC is limited to Mistere Falcon 20-ES aircraft ser...</td>
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<tr>
<td>EASA.A.S.03538</td>
<td>TATENHILL AVIATION Ltd.</td>
<td></td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>CESSNA 525</td>
<td>Dual Garmin GPS/NAV/COM Upgrade.</td>
<td>Mod. Ref. TAL-TAD 065/06MDL 05058, Rev.A/AFM Supplement - FMS.050508 Rev. A.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<tr>
<td>EASA.A.S.03539</td>
<td>AIRLEC AIR ESPACE</td>
<td></td>
<td>SA226-T(B)</td>
<td></td>
<td>Stretcher Installation: New stretchers installation put in the aircraft with storage place and O2 bottle space.</td>
<td>Airplane Flight Manual Supplement &quot;Supplement Manuel de Vol SA226-T(B) - Installation civienne&quot; Edition 1, Juillet 2008-Aircraft Maintenance Manual Supplement &quot;Supplement Manuel de Maintenance SA-226(B) - Installation civienne&quot; Edition 1, Juillet 2008.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>STC Ref.</td>
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<td>EASA.A.S.03544</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>BEECH 400</td>
<td>4</td>
<td>BEECH 400, BEECH 400A</td>
<td>Mode S ELT/ENI Upgrade of existing transponder system.</td>
<td>Engineering order. Doc. No. 2007-22-2, Rev. B or latest approved revision.AMS, Doc. No. 2007-22-8, Rev. B or latest approved revision.</td>
<td>2042065-ED issue 2 AFMS 2042065-AM issue 1.</td>
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<td>EASA.A.S.03545</td>
<td>CESSNA DUESSELDORF CITATIONBERVIC E CENTER GmbH</td>
<td>PILATUS PC-12</td>
<td>4</td>
<td>PILATUS PC-12, PILATUS PC-1247</td>
<td>Installation of Honeywell EGPW5 MK VI by installing a Honeywell EGPW5S MK-VI.</td>
<td>Engineering order. Doc. No. 2007-22-2, Rev. B or latest approved revision.AMS, Doc. No. 2007-22-8, Rev. B or latest approved revision.</td>
<td>10/06/2008</td>
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<td>EASA.A.S.03547</td>
<td>J &amp; C ENGINEERING SERVICES Ltd.</td>
<td>REV 1</td>
<td>2</td>
<td>R2160, R2160D, R2160I, R2120U</td>
<td>Removal of VFR Limitation from AFM Approved Operation Section.</td>
<td>Engineering Report J/C/ER/1444 Issue 2.0 Flight Manual Supplements J/C FMS/1444 Issue 1.1, J/C FMS/1444-1 Issue 1.0, J/C FMS/1444-2 Issue 1.0, J/C FMS/1444-3 Issue 1.0</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>EASA.A.S.03553</td>
<td>CAA UK RESPONSIBILITY (STC TWENTY ONE LIMITED)</td>
<td>BOEING 737-300</td>
<td>2</td>
<td>Carriage of Horses &amp; Grooms</td>
<td>Installation.</td>
<td>1. Master Document List S21. MDL-024 Issue 5°. Airplane Flight Manual Supplement S21, FMS-0477 Issue 1°. Instructions for Continued Airworthiness S21. TEC-192 Issue 1° or later approved revisions of those documents.</td>
<td>1. This design change is applicable to Boeing 737-300 aircraft converted under FAA STC SA266590, SA266590, and SA297050 &amp; SA297150 as previously been validated by the UK CAA under Airworthiness Approval Note AAN 24984.2. The certification basis has been amended with the following CRIs: 0-D.</td>
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<tr>
<td>EASA.A.S.03554</td>
<td>LUFTHANSA TECHNIK AG</td>
<td>AIRBUS A318-112</td>
<td>2</td>
<td>A318-112 - ELITE - Installation of a VIP cabin for a maximum of 19 passengers for commercial and non-commercial operation. The design details are described in General Description S-00-08/731.</td>
<td>Installation of L3 FDR1000; Collins EGPW5 Mark VI; Honeywell CAS 67 &amp; Dual MST 67A Mode S Diversity Transponder; Airex 406-2 ELT</td>
<td>• Supplemental Type Definition Document, ref. Doc. No. STDD00-08/03M, issue 01, dated 27 January 2009, or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification is b...</td>
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<tr>
<td>EASA.A.S.03556</td>
<td>AIR ALLIANCE GmbH</td>
<td>REV. 1</td>
<td>2</td>
<td>300</td>
<td>Installation of L3 FDR1000; Collins EGPW5 Mark VI; Honeywell CAS 67 &amp; Dual MST 67A Mode S Diversity Transponder; Airex 406-2 ELT</td>
<td>The installation of this equipment shall be done in accordance with Air Alliance Engineering Order 2008_014_A242E_0000, Revision 00, Issue 25-06-2008, or later EASA approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
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<td>EASA.A.S.03558</td>
<td>JET AVIATION VIENNA GMBH</td>
<td>550</td>
<td></td>
<td></td>
<td>Upgrade of SSR Transponder System to Enhanced Surveillance.</td>
<td>Certification Report CS50-ARS34-003, Rev. 1, 23.06.2008 and 02.05.2008</td>
<td></td>
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<tr>
<td>EASA.A.S.03560</td>
<td>DASSAULT FALCON SERVICE</td>
<td></td>
<td></td>
<td></td>
<td>Falcon 900C and F900EX: Incorporation of Runway Awareness and Advisory System (RAAS) Function software version 224-224 on EGPWS Honeywell Mk V</td>
<td>DFS Master Document List DFS-DE-17-F9-0641 Issue A1 or later approved revisions DFS Change Approval Sheet DFS0541-CAS Issue A1 or later approved revisions DFS AFM Supplement F900-027 Original Issue (dated 09th June 2008) or later approved revisions DFS Maintenance manual Supplement SM5 31-05-11-A2</td>
<td>As per applicability defined on DFS Change Approval Sheet DFS9541-CAS Issue A1 or later approved revisions this STC is limited to those Falcon 900C and Falcon 900EX aircraft previously equipped with EGPWS Honeywell Mk V Soft. 212-212 with terrain display on EFS-Honeywell DU-141 with BZ control...</td>
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<tr>
<td>EASA.A.S.03561</td>
<td>AERODATA AG</td>
<td>PC-12</td>
<td></td>
<td></td>
<td>Modification as surveillance aircraft</td>
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<tr>
<td>EASA.A.S.03563</td>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>114, 114A, 114B, 114TC, 112B, TC, TCA, 114A, B, TC</td>
<td></td>
<td></td>
<td>Replacement of existing COM 1 / 2 and NAV 1 / 2 systems by Garmin GNS 530 (GNS 1) and Gamin GNS 430 (GNS 2)</td>
<td>EASA approved Engineering Order, Doc. No. 2048711-EO, Issue1, dated 21-Apr-2008, or later approved Issue</td>
<td>This change is limited to the configuration of aircraft / equipment type as listed under Item 1. of EASA approved Engineering Order, Doc. No. 2048711-EO, Issue 1, dated 21-Apr-2008 or later approved issue.</td>
</tr>
<tr>
<td>EASA.A.S.03565</td>
<td>FR AVIATION LTD/D/IA COBHAM AVIATION SERVICES UK</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>FAN JET FALCON SERIES D AND E</td>
<td>FAN JET FALCON BASIC</td>
<td>P-RNAV Compliance Demonstration, as per FR Aviation Design Change Statement DC-0133</td>
<td>- FR Aviation Design Change Statement DC-0133 Issue 1.1 - FR Aviation Flight Manual Supplement FRAFPM108, Revision 10.1, Instructions for Continued Airworthiness ICA-053, Revision A or later EASA Approved revisions of these documents.</td>
<td>1.  Design Change applicable to those Fan Jet Falcon Basic and Fan Jet Falcon series D and Eos which FR Aviation STC EASAAS.0510 (Rockwell Collins Proline 4) have been previously embodied. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<tr>
<td>EASA.A.S.03568</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BOMBARDE BD-700-1A11</td>
<td></td>
<td>Introduction of Passenger Cabin Interior.</td>
<td>- STC Twenty One Ltd Master Document List S21.MDL-0532 dated 09 December 2008 or later approved revision. - STC Twenty One Ltd Flight Manual Supplement S21.FMS-048 Issue 1 or later approved revision.</td>
<td>1.  Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third party...</td>
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<tr>
<td>E.I.S. AIRCRAFT GmbH</td>
<td>SOCATA TBM 700</td>
<td>BEECH C90KING AIR SEECH C90 A</td>
<td>Installation of dual Rockwell/Colliers TDF 94D diversity mode-s transponder.</td>
<td>Prior to installation of this modification the installer must determine that the interoperability between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons ...</td>
<td>19/12/2008</td>
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<td>CESSNA DUSSIELODORF CENTER GmbH</td>
<td>CESSNA 525 CESSNA 525</td>
<td>Installation of GARMIN GPS 400 (w) or 500 (w)</td>
<td>EASA approved Engineering Data, Doc. No. 2008-2, Revision B, dated 11-Feb-2009, or later approved issue. EASA approved Airplane Flight Manual Supplement, Doc. No. 2008 2 A8, Revision D, dated 03-Mar-2009, or later approved issue. Maintenance Manual Supplement, Doc. No. 2008-2-10, Revision ...</td>
<td>1. For approved Software Versions refer to Airplane Flight Manual, Section 9. Limitations 2. Prior to installation of this modification the installer must determine that the interoperability between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons ...</td>
<td>28/04/2009</td>
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<td>AVIONITEC Ltd.</td>
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<td>114</td>
<td>Avionics Retrofit Removed Equipment.- KMA 20 Audio Panel- Isocom Intercom Intercom- Dual RX 10i COM/NAV- Dual TX 255 NAV 2 Indicator-GPS 1, Garmin GPS 100 Ind. Antenna- OEM vacuum Directional Gyro- KT 78A Transponder- Narco ELT 19New Installed Equipment.- GMA 340 Audio Panel- Dual GNS 430 COM/NAV/GPS with GA 5 ...</td>
<td>Master Document List, Doc. No. PE8100-017, Rev. 1 or latest approved Revision- AFMS Dual GNS 430, Doc. No. PE8100-011, Original Issue or latest approved Revision- AFMS SN2/000 HSG, Doc. No. PE8100-012, Original Issue or latest approved Revision- AFMS EX500 MFQ, Doc. No. PE8100-013, Original Issue or ...</td>
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<td>EASA.A.S.03580</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
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<td>HAWKER HS 125-700 SERIES HAWKER HS 125-600 SERIES BAE 125-800 SERIES</td>
<td>Installation of a Digital Flight Data Recorder.</td>
<td>Installed in accordance with 2047003-EO Engineering Order 2 or later approved revision.Operated in accordance with 2047003-AM AFM supplemental issue 1 or later approved revision. Maintained in accordance with 2047003-CA Instructions for Continued Airworthiness Issue 1 or later approved revision.</td>
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<td>EASA.A.S.03581</td>
<td>SOCIETE AIR FRANCE S.A.</td>
<td>REV 1</td>
<td>BOEING 777-300 BOEING 777-200</td>
<td>ADS-B Out Activation.</td>
<td>AFMS: B777-07-023CA B777_0 7.023_rev0_11JU08 None.</td>
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<td>EASA.A.S.03582</td>
<td>S.E.A.M. SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td></td>
<td>PIPER PA-44-180</td>
<td>Modification d'Installation B-RNAV : pose Annonciateur DWA 1000</td>
<td>1) Fiche de Modification SEAM 208058FACE/SEAM Edition 1 Revision 3 du 22/05/2008/Annexe C1 - Plan d'implantation des équipements +Annexe C2 - Plan électrique Edition 1 Révision1 du 28/07/2008/Annexe C3 - Supplement No1 au manuel de vol/Annexe...</td>
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<td>EASA.A.S.03587</td>
<td>AUSTRIAN AIRLINES AG</td>
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<td>F28 MARK 0070, F28 MARK 0100</td>
<td>Installation of mechanically timed water faucet in Fokker F28 Mark 0070, Fokker F28 Mark 0100.</td>
<td>Supplemental Type Definition Report Nr STD-FFKR-38-0001-02.</td>
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<td>AIRBUS S.A.S.</td>
<td>A320-214</td>
<td>Airbus Corporate Jet Centre Modification CJ3036 Issue 1 - AFM Supplement - VIP Interior: Shining Star No. 2</td>
<td>Modification Approval Sheet ref.: CJ3036 Issue 1, dated 30th June 2008 AFM Supplement: FM-8.MOD CJ3013, dated 06/06/08</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/06/2008</td>
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<td>EASA.A.S.03590</td>
<td>STC TWENTY ONE LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>BOEING 767-200</td>
<td>Introduction of Change 7.0 ACAS II and Dual Mode S STC 21 Master Data List SI21.MDL-0543 Initial Issue, dated 12/06/2008</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/06/2008</td>
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<td>EASA.A.S.03592</td>
<td>2 EXCEL DESIGN LIMITED</td>
<td>A320-232</td>
<td>A320 in Flight Entertainment – IFE – Upgrade</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>29/10/2008</td>
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<td>EASA.A.S.03597</td>
<td>JET AVIATION AG, BASEL</td>
<td>GULFSTREAM IV</td>
<td>FMS Upgrade Installation new EHSI 4000, NZ2000 SW 6.0 Installation, NewCDU C2-300.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>18/05/2009</td>
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<td>EASA.A.S.03598</td>
<td>JET AVIATION AG, BASEL</td>
<td>GULFSTREAM IV</td>
<td>Wireless LAN (WLAN) Activation.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>28/06/2008</td>
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<td>EASA.A.S.03599</td>
<td>SCHEDEL ENGINEERING Ltd.</td>
<td>SCHEMPP CIRRUS</td>
<td>Standard Cirrus: Weight of nonlifting parts increase with winglets Doc. No. 07-278-006, rev. 2, Doc. No. 07-278-006, rev. 3</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.A.S.03603</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>M7 AEROSPACE SA26-AT</td>
<td>Installation of 5-blade MT-Propeller Model MTV-27-1-E-C-F(R)(G)CFR210-38a propellers.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.A.S.03604</td>
<td>MT-PROPELLER ENTWICKLUNG GmbH</td>
<td>PIPER PA-42-720 PIPER PA-42</td>
<td>Installation of 4-blade MT-Propeller Model MTV-16-1-E-C-F-R/PYCFR230-37b propellers.</td>
<td>According to MT-Propeller Master Document List Doc. No. E-1682 and MT-Propeller Installation Instruction Doc. No. E-1671, original issue, or later Revisions To be operated in accordance with MT-Propeller AFM &amp; Document No. E-1670, original issue, or later approved Revisions To be maintained i.a.w. MT-...</td>
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<td>EASA.A.S.03619</td>
<td>JET AVIATION AG, BASEL</td>
<td>AIRBUS A319-133</td>
<td>Wireless Local Area Network (WLAN) Activation</td>
<td>Reference Data List, ref. Doc. BA31923262-001, Rev. A, dated 13.08.08 Certification Compliance Sheet, ref. Doc. BA31923262-012-01, Rev. A, dated 12.08.08 Airplane Flight Manual Supplement, ref. BA31923262-021-01, Rev. IR, EASA approval date 13.08.08 Instruction for Continued Airworthiness...</td>
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<td>EASA.A.S.03625</td>
<td>DGAC STC. 180-SF-0227</td>
<td>A320 SERIES</td>
<td>Installation of a cabin VIP (Aeroservices) in accordance with the Service Bulletin SOG-WD-00-00-01 (or later approved revision)</td>
<td>Supplement no 180-SF-0227 to data sheet no 180Service Bulletin SOG-WD-00-00-01 (or later approved revision)</td>
<td>04/07/2008</td>
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## List of EASA Supplemental Type Certificates

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<tr>
<td>AIRBUS A319 Corporate</td>
<td>Cabin Conversion of one A319 Corporate Jet (A319 CJ)</td>
<td>Type Certificate Data Sheet no 180 and its supplement no 180-SF-0212 with deviations identified and listed in the Certification Basis document, CRT G-4102 issue 3, dated November 2003</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>07/12/2008</td>
<td>Active</td>
<td>1012122</td>
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<td>AVIONICS INTEGRATION &amp; ENGINEERING &amp; RPRATION AG (AIEC)</td>
<td>Modification of installation transpondeur Mode S: dépose 2 transpondeurs XUPR 755 + Boîte de commande + 2 antennes ATC / Pose 2 transpondeurs GTX 330D + 4 antennes ATC</td>
<td>Modification effectué selon le document S.E.A.M « Fiche démonstration 2125/08 » ; édition 1, révision 0 ; or later EASA approved revision Aircraft Flight Manual Supplement document SEAM/SM/2125/08, révision 0, du 30/09/2008, or later EASA approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td>Modification of installation transpondeur Mode S: dépose 2 transpondeurs XUPR 755 + Boîte de commande + 2 antennes ATC / Pose 2 transpondeurs GTX 330D + 4 antennes ATC</td>
<td>Modification effectué selon le document S.E.A.M « Fiche démonstration 2125/08 » ; édition 1, révision 0 ; or later EASA approved revision Aircraft Flight Manual Supplement document SEAM/SM/2125/08, révision 0, du 30/09/2008, or later EASA approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>20/02/2009</td>
<td>Active</td>
<td>1012125</td>
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<td>BEECH 200</td>
<td>Modification of installation transpondeur Mode S: dépose 2 transpondeurs XUPR 755 + Boîte de commande + 2 antennes ATC / Pose 2 transpondeurs GTX 330D + 4 antennes ATC</td>
<td>Modification effectué selon le document S.E.A.M « Fiche démonstration 2125/08 » ; édition 1, révision 0 ; or later EASA approved revision Aircraft Flight Manual Supplement document SEAM/SM/2125/08, révision 0, du 30/09/2008, or later EASA approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>17/04/2009</td>
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<td>LUFTHANSA TECHNIK AG</td>
<td>Modification of installation transpondeur Mode S: dépose 2 transpondeurs XUPR 755 + Boîte de commande + 2 antennes ATC / Pose 2 transpondeurs GTX 330D + 4 antennes ATC</td>
<td>Modification effectué selon le document S.E.A.M « Fiche démonstration 2125/08 » ; édition 1, révision 0 ; or later EASA approved revision Aircraft Flight Manual Supplement document SEAM/SM/2125/08, révision 0, du 30/09/2008, or later EASA approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<tr>
<td>AIRBUS A318-112 - ELITE - Project Tanker</td>
<td>Modification of installation transpondeur Mode S: dépose 2 transpondeurs XUPR 755 + Boîte de commande + 2 antennes ATC / Pose 2 transpondeurs GTX 330D + 4 antennes ATC</td>
<td>Modification effectué selon le document S.E.A.M « Fiche démonstration 2125/08 » ; édition 1, révision 0 ; or later EASA approved revision Aircraft Flight Manual Supplement document SEAM/SM/2125/08, révision 0, du 30/09/2008, or later EASA approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>16/06/2009</td>
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<td>AIR Alpha Maintenance A/S</td>
<td>Flight Inspection System antenna and equipment interface installation.</td>
<td>Installation per Air Alpha Accomplishment Instruction AE1-E0-B200-25-122 Rev 0 dated 15 October 2008 and Master Drawing List – Antenna &amp; Equipment Interface Installation EO-B200-25-122 dated 6 Nov 2008 – Operations per Air Alpha Flight Manual Supplement FAI.EO-B200-25-122 Rev.00 dated 15 Apr 2.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>03/12/2008</td>
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<td>EASA.A.S.03636</td>
<td>COMPAGNIA GENERALE RIPRESSEAREES p.A. - COR S.p.A.</td>
<td>PARTENAVIA P68</td>
<td>PARTENAVIA P68</td>
<td>Upgrade of Mode S transponders for enhanced surveillance.</td>
<td>-- Installation in accordance with Lees Avionics Ltd. Change number/H/MD/ 994- AFMS Lees Avionics Ltd. Supplement 994001, Issue 1, or later EASA approved revision</td>
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<td>EASA.A.S.03643</td>
<td>AIR ALLIANCE GmbH</td>
<td>REV. 1</td>
<td>525, 525A</td>
<td>Installation of Dual GTX 330D as Replacement of Dual KT 70 or KT 71</td>
<td>AirAlliance Master Document List 2008_036_A1WAA Form 2, in latest approved Revision To be operated in accordance with AFM Supplement Document No 2008_036_A1W1_506, Rev 0, or later approved Revisions To be maintained i.a.w. ICA Document No 2008_036_A1W1_507, Rev 1, or later approved Revisions</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>JET AVIATION AG, BASEL</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Ansat Iridium Antenna P/N. 015-10306-001 Installation</td>
<td>Jet Aviation Declaration of Compliance dated 12 August 2008Jet Aviation Master Data List No. B7REPCUS1044-001, Revision A dated 12 August 2008</td>
<td>Jet Aviation Declaration of Compliance dated 12 August 2008Jet Aviation Master Data List No. B7REPCUS1044-001, Revision A dated 12 August 2008</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>AVIONIK STRAUBING ENTWICKLINGS GmbH</td>
<td>501, 551</td>
<td>Installation of COM/NAV/GPS Garmin GNS430W (AW) or COM/NAV/GPS/ GNS530W (AW).</td>
<td>1. According to Avionik Straubing Master Document List Doc. No. ASR-2008-065-MDL-01-00 Rev. 00, or later Revisions. 2. To be operated in accordance with approved Avionik Straubing AFM-S Document No. ASR-2008-065-FMS -05-00, Rev. 00, or later approved Revisions. 3. To be maintained i.a.w. Av...</td>
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<td>KUERZI AVIATION AG</td>
<td>PILATUS PC-12/47E</td>
<td>FDR and CVR Installation.</td>
<td>1. MDL-200830253 Rev. A or later approved revision2. FMS-200830253 Rev. N/C or later approved revision3. ICA-200830253 Rev. A or later approved revision</td>
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<td>EASA.A.S.03653</td>
<td>CESSNA DUESSELDORF CITATIONSERVICE CENTER GmbH</td>
<td>CESSNA CITATION CS25</td>
<td>Installation of AIRCELL Iridium Dual Patch Satcom Antenna.</td>
<td>EASA approved Engineering Data, Doc. No. 2008-3, Revision B, dated 27-Apr-2009, or later approved Issue; Maintenance Manual Supplement, Doc. No. 2008-3-10, Revision B, dated 27-Apr-2009, or later approved Issue.</td>
<td>1. Antenna installation only. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA.A.S.03655</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td>BEECH B300 BEECH B200 BEECH 300</td>
<td>Be 'Mode S / Flight ID' capable Collins TDR-94(D) transponders.</td>
<td>The installation will be made in accordance with Business and Commuter Aircraft document BCA108-00, Revision 0, as of 14.08.2008 or later EASA approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA.A.S.03658</td>
<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>35A</td>
<td>Solid State Flight Data Recorder SSFDR Installation (L-3)</td>
<td>As detailed in Certification Program AG748-080, rev. IR (or later approved revisions).</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>AERO-DIENST GMBH &amp; CO. KG</td>
<td>35A</td>
<td>Cockpit Voice Recorder CVR-30B Installation (UASC).</td>
<td>As detailed in Certification Program AG751-080, rev. IR (or later approved revisions).</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA.A.S.03660</td>
<td>AERODATA AG</td>
<td>31A</td>
<td>Installation of Aerodata Flight Inspection System (FIS)</td>
<td>As detailed in Master Document List MDC-A/C-139, rev. 1 (or later approved revisions).</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by third person...</td>
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<td>JET AVIONICS AVIATION &amp; MARINESERVICE CENTER S.r.l.</td>
<td>CESSNA CITATION 501</td>
<td>Installation of Mode S Transponder.</td>
<td>According to Jet Avionics Master Document List Doc. No. BI005-08-01 and Jet Avionics Installation Bulletin Doc. No. BI015-08-03, original issue, or later Revisions. To be operated in accordance with Jet Avionics AFM-M Document No. BI005-08-18, original issue, or later approved Revisions.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by th...</td>
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<td>STC Holder</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>SABENA TECHNICS BGC</td>
<td>SN 601</td>
<td>Finish Material Change on Aerospatiale SN 601 'Corvette' This STC concerns the change of finish materials in the cabin such as interior linings, seat foam and cover and floor.</td>
<td>Master Data List ref. 4187-01-A-2103-R1, or any later approved revisions of this document, except for referenced document which are approved separately, as ICA for instance Instructions for Continued Airworthiness, ref: 4187-01-A-2001-R00</td>
<td>This STC is limited to the SN 601 'Corvette' MSNs : 11, 13, 30, 31, and 32 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness...</td>
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<td>BUSINESS AND COMPUTER AIRCRAFT</td>
<td>BEECH F90 BEECH E90 BEECH C90</td>
<td>BCA 110 Rev. 3 - Garmin GNS330WA (GNS-530WA or GNS-430WA) and Sandel SN4500 installation for B-RNAV, P-RNAV and GNSS approaches.</td>
<td>AFM Supplement BCA110-07 rev. 2, dated 28 November 2008, AIM supplement BCA110-06 rev. 5, dated 22 August 2008.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product...</td>
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<td>AEROSPACE DESIGN AND ENGINEERING NSULTANTS Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>Introduction of 12C241Y Passenger Cabin Configuration and Associated Floor Structure Changes.</td>
<td>AeroDEC Modification Summary MOD-0477, Revision B</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product...</td>
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<td>S.E.A.M SERVICE ELECTRONIC AVIATION MARINES A.R.L.</td>
<td>CESSNA 182R</td>
<td>MODIFICATION D'INSTALLATION B-RNAV avec couplage au pilote automatique</td>
<td>Supplement manuel de vol SEAM/SMV/212/07 Rev. 0</td>
<td>Limitation to Cessna 182R S/N 182-68316</td>
<td>30/03/2009</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
<td>B. 441 A. 421, 421A, 421B, 421C, 42a A. 411, 411A, 414, 414A A. 402, 403A, 402B, 402C A. 401, 401A, 401B</td>
<td>Installation of Single or Dual Garmin G600 Avionics System and dual Garmin GNS435/530/530W GPS/COM/NAV</td>
<td>-Project Description: ASR-2008-104-PRD-04-00, Rev. 0, Instructions for Installation: ASR-2008-104-PRD-04-00, IPI-10-00, Rev. 0, Instructions for Continued Airworthiness: ASR-2008-104-A-CA-11-00, Rev. 0, Flight Manual Supplement: ASR-2008-104-FMS-05-00, Rev. 0</td>
<td>1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2) The installation of this modification by third persons...</td>
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<td>SOCIETE AIR FRANCE S.A.</td>
<td>BOEING 747-400</td>
<td>ADS-B Out Installation and Activation as per Air France Service Bulletin 8247-08-004-34-001 rev. 0 or later approved revisions.</td>
<td>AFMS 8747-08-004 Airplane Flight Manual Supplement for 8747-400 ADS-B Out Installation.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2) The installation of this modification by...</td>
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<td>2 EXCEL DESIGN LIMITED</td>
<td>CAT UK RESPONSIBILITY</td>
<td>Banbury Modification 001 - FLR Installation.</td>
<td>JN147 ADC17 Issue 5 – Master Data List.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third pe...</td>
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<td>Limitations/Conditions</td>
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<td>EASA.A.S.03684</td>
<td>GAMA ENGINEERING LTD.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>HAWKER 700</td>
<td>Introduction of a Sandel ST3400 Class A terrain awareness warning system.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Flight Manual Supplement TEN/3400-01 Form 1, dated 04/02/09</td>
<td>12/02/2008</td>
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<td>EASA.A.S.03687</td>
<td>BELL TEXTRON PRAGUE a.s.</td>
<td>L-410 UVP-E</td>
<td>Installation of EGPWS Mark VI.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Flight Manual Supplements - No FM 34 41/01/01 (Multi-Functional Display XMD 540), dated Nov 11/08 - No FM 34 43/01/01 (EGPWS Mark VII), dated Nov 11/08 - No FM 34 55/01/01 (Radar Altimeter KRA 405), dated Nov 11/08 - Engineering Bulletin - No: EB 164, dated 11.11.2008</td>
<td>None</td>
<td>24/11/2008</td>
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<td>EASA.A.S.03696</td>
<td>TENENCIA Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>EMBRAER EMB-135 SERIES</td>
<td>EMB 135 Electronic Flight Bag installation</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>TEN 582 Issue 2 dtl. 13.5.09 += Flight Manual Supplement TEN/EMB135/01 Issue 2 **) or later approved revision</td>
<td>29/05/2009</td>
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<td>EASA.A.S.03697</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERINGCO CORPORATION AG (AIEC)</td>
<td>BOMBARDEIR CL-600-2B16</td>
<td>P-RNAV Compliance</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Data List No. AC780-049 AFM Supplement Change Sheet No. AC780-071</td>
<td>None</td>
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<td>2 EXCEL DESIGN LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>DC-9-32</td>
<td>119Y LOPA (Removal of one row of seats to improve access to type III exits to meet local retroactive requirements).</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Master Data List MDL/JH395, issue 1, dated 14 August 2008.</td>
<td>20/06/2008</td>
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<td>EASA.A.S.03700</td>
<td>ICELANDAIR TECHNICAL SERVICES</td>
<td>BOEING 757-200</td>
<td>Reconfigure seating layout to 52/10 Pax arrangement (VIP &amp; Lounge)</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Change/Repair Approval, CAA-757-2520-276, Revisions B, C, dated 11/09 and CAA-757-2520-276, Revision CR, dated 30/09/2008</td>
<td>None</td>
<td>15/12/2008</td>
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<td>EASA.A.S.03701</td>
<td>AEROTEC SARL</td>
<td>TB 20</td>
<td>Installation of L3 Communication Stormscope WX500, Garmin audio system GMA 340 and Garmin GNS 530 W for IFR B-RNAV en-route navigation.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>AE84159/N, revision 0, dated 04/02/09 - Aerotec Master Data List no: AE 841 MDL, revision 1, dated 02/03/09</td>
<td>This installation is limited to aircraft TB 20 s/n 1801 which must be equipped with a third radio-communication or radio-navigation means independent from GNS 430 and GNS 530 equipment.</td>
<td>07/05/2009</td>
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### List of EASA Supplemental Type Certificates

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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
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<td>EASA.A.S.03707</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SIGA)</td>
<td>10015259</td>
<td>GAMA</td>
<td>BEECH 1900</td>
<td>Installation of a galley.</td>
<td>Description: Installation sur avion Beech 1900 d'une Antenna S&amp;S.</td>
<td></td>
<td>Prior to installation of this modification the installer must determine that the modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons.</td>
<td>16/12/2009</td>
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<td>EASA.A.S.03720</td>
<td>CHRISTEN-ARITHEK</td>
<td>REV 1</td>
<td>CESSNA 210</td>
<td>Avionics Upgradability removing the existing COM/NANN 1 + 2. DME: Mode A/C Transponder- Audio Panel/Mode S Transponder- GMA 340 Audio Panel/Mode S Transponder- DORNIER 228-300 MODE 2</td>
<td>Master Data List, Doc. No. 208.2027, Revision 01, or later approved revision. AFMS GNS 530W COM/NAV/GPS 1, Doc. No. 208.2027, Revision 01, or later approved revision. AFMS GNS 430W COM/NAV/GPS 2, Doc. No. 208.2027, Revision 01, or later approved revision. ICA, Doc. No. 208.2027, Revision 01.</td>
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<td>EASA.A.S.03723</td>
<td>GAMA ENGINEERING LTD</td>
<td>REV 1</td>
<td>CESSNA 310</td>
<td>Introduction of EXP5000 electronic primary flying display and dual GNS 430W GPS navigation/communication units.</td>
<td>Lees Avionics Limited Change HLMOD 1032 Flight Manual Supplement AFMS1032-01 Issue 1</td>
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<td>27/07/2009</td>
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<td>DASSAULT FALCON SERVICE</td>
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<td>FALCON 900</td>
<td>Installation of two Honeywell 12 or 24 Channels GPS sensors(GNSSU)</td>
<td>- Master Document List DFS-DE-17-F9-9119 Issue HP-Airplane Flight Manual Supplement DFS Supplement P900-029 OriginalversionOr later EASA approved revisions of these documents.</td>
<td>1. Prior to installation of this modification the installermust determine that theinterrelationship between this modification and any other previouslyinstalled modificationwill introducenoe adverse effect upon the airworthiness of the product.2. The installation of this modification by third persons ist...</td>
<td>08/07/2009</td>
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<td>EASA.A.S.03729</td>
<td>DASSAULT FALCON SERVICE</td>
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<td>FALCON 900</td>
<td>Dual installation of Garmin GNS430W/GNS430AW/GNS 400 GNS530W/GNS530W/GNS530W</td>
<td>Installation must be in accordance with Master Document List EngineeringOrder No. ASR2008-125-MDL-01 Amend. 00 dated Jan. 2009 or later ApprovedRevisions.Operati...</td>
<td>02/06/2009</td>
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<td>AVIONIK STRAUBING ENTWICKLUNGS GmbH</td>
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<td>PA-34-220T PA34-200, PA34-200T</td>
<td>One or two three-places Sofa installation in the aft part of the Cabin.</td>
<td>DFS Master Document List DFS-DE-17-F9-9549 Issue A1 or later approved revisions.DFS Change Approval Sheet (DFS5949-CAS) Issue A1 or later approved revisions.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by third persons...</td>
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<td>BOEING MD 11</td>
<td>Auxiliary Fuel Tank Removal, B737-400.</td>
<td>Master Drawing MAP184101 (a. D.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>16/02/2009</td>
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<td>TENENCIA Ltd.</td>
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<td>BOEING 757</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/03/2009</td>
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<td>AERODATA AG</td>
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<td>AERODATA AG</td>
<td>Installation of an Aerodata Flight Inspection System AD-AFIS-111</td>
<td>L. EASA approved Master Document List, Doc. No. MOD-AF/144, Revision 1, dated 10-Mar-2009, or later approved revision. EASA approved Airplane Flight Manual Supplement, Doc. No. AFM 104, Revision 0, dated 20-Nov-2008, or later approved revision. EASA accepted Maintenance Manual Supplement, D.</td>
<td>1. Airplane must be equipped as stated in Chapter 1.3.1 of EASA approved Aerodata Certification Program, Doc. No. CP.B3890-13.01 Revision 0, dated 12-Jan-2009, or later approved revision.2. Prior to installation of this modification the installer must determine that the internal relationship between...</td>
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<td>STC TWENTY ONE LIMITED</td>
<td>737-300</td>
<td>CAA UK RESPONSIBILITY</td>
<td>AERODATA AG</td>
<td>Carriage of Horses &amp; Grooms</td>
<td>- Master Document List S21.MDL-0012 Issue 1., Airlne Flight Manual Supplement S21.FMS-0051 Issue 1., Instructions for Continued Airworthiness S21.TEC-003 Issue 10, later approved revisions of those documents.</td>
<td>1. This design change is applicable to Boeing 737-300 aircraft converted under FAA STC SA2968SO, SA2969SO, and SA2970SO previously validated by the UK CAA under Airworthiness Approval Note AAN 24864.</td>
<td>18/06/2009</td>
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<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>A300F4-605ST</td>
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<td>AIRBUS TRANSPORT INTERNATIONAL SNC</td>
<td>Installation of pressure seals on Frame (FR) 10.7. This STC consists in installing two pressure seals on FR10.7 in order to allow future avionics installations.</td>
<td>Change Approval Sheet, ref. STS3MDD011 Issue 1 (Any later approved revision).</td>
<td>1) Prior to installation of this modification the installer must determine that the internal relationship between this modification and any other previous installed modification will introduce no adverse effect upon the airworthiness of the product. This Certificate shall remain valid unless otherwise surm...</td>
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<td>AIR ALLIANCE GmbH</td>
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<td>PIPER PA-31T2</td>
<td>Camera Installation Floor Cut Out</td>
<td>According to AirAlliance Master Document List Doc. No. 2008_058_AEAA AA Form 2, Rev. 0, or later approved Revisions and/or AllInAir Installation Instructions 2008_058_AEAA 500, Rev. 0, or later Revisions To be operated in accordance with AirAlliance AFM-S Document No. 2008_058_AEAA 500, Rev. 0, or later...</td>
<td>1. Prior to installation of this modification the installer must determine that the internal relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification...</td>
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<td>AERODISA S.L.</td>
<td>FMS Upgrade to PRNAV Compliance (JASIC).</td>
<td>As detailed in Certification Program AD794-080, rev. IR (or later approved revisions).</td>
<td>1. Prior to installation of this modification the installer must determine that the internal relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by...</td>
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<td>AERODISA S.L.</td>
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<td>MYSTERE FALCON 900 (MF900)</td>
<td>Upgrade of dual Mode S SSR Transponder installation to provide both Elementary Mode S Surveillance Flight ID and Enhanced Mode S Surveillance capability (EHS Capable).</td>
<td>See Aerodata documents - Master Data List AE2D008-06_F000_EHS MDM Revision 04, Flight Manual Supplement AE2D008-09_F000_EHS D03 Revision 05, Maintenance Supplement Manual AE2D008-09_F000_EHS D02 Revision 01, Installation Manual AE2D008-09_F000_EHS D01 Revision 03 Or later EASA approved revisions</td>
<td>1. Prior to installation of this modification the installer must determine that the internal relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by...</td>
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<td>GULFSTREAM</td>
<td>Coupling of GPS Trimble 2101 I/O</td>
<td>Aircraft Flight Manual Supplement (Allegato N) 1st Edition 2001</td>
<td>Aircraft Flight Manual No. 700-0202-800-000 Rev. (R) dated 17.06.2002 and later approved revisions. This Certificate shall remain valid unless otherwise surrendered or revoked.</td>
<td>Doc. DGN 400-001-000 a successive revision approvals.</td>
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<td>VITROCISET S.p.A.</td>
<td>ENAC STC CS-071</td>
<td>Installation of EGPWS Universal Avionica Systems with GPS 1000 and/or FDM+ (=) + Note 1: This STC reflects the transfer of ENAC STC CS-071 from LEAT S.r.l. to VITROCISET S.p.A. This STC supersedes ENAC STC CS-071.</td>
<td>Master Drawing List No. 700-0202-000-000 IR dated 17.06.2002 and later approved revisions.</td>
<td>Included in the Supplemental Flight Manual No. 700-0202-800-000 Rev. (R) dated 17.06.2002 and later approved revisions. This Certificate shall remain valid unless otherwise surrendered or revoked.</td>
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<td>VITROCISET S.p.A.</td>
<td>ENAC STC CS-070</td>
<td>Upgrade FDR installation with installation of Digital SSDFR type F1000. Note:1. This STC reflects the transfer of ENAC STC CS-070 from LEAT S.r.l. to VITROCISET S.p.A.2. This STC supersedes ENAC STC CS-070.</td>
<td>Master Drawing List No. 600-0402-000-000 IR dated 17.06.2002 and later approved revisions.</td>
<td>Included in the Supplemental Flight Manual No. 900-0402-800-000 Rev. (R) dated 17.06.2002 and later approved revisions. This Certificate shall remain valid unless otherwise surrendered or revoked.</td>
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<td>VITROCISET S.p.A.</td>
<td>ENAC STC CS-062</td>
<td>Coupling of GPS Trimble 2101 I/O Approach Plus with Autopilot SP2 550 Note 1: This STC reflects the transfer of ENAC STC CS-062 from LEAT S.r.l. to VITROCISET S.p.A. This STC supersedes ENAC STC CS-062.</td>
<td>MDL DWG No. 100-0502-000-000 dated 24.04.2002 and later approved revisions.</td>
<td>Included in the Supplemental Flight Manual No. 100-0502-000-000 dated 05.03.2002 and later approved revisions. This Certificate shall remain valid unless otherwise surrendered or revoked.</td>
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<td>VITROCISET S.p.A.</td>
<td>ENAC STC CS-091</td>
<td>Coupling of GPS GARMIX 155XL with Autopilot Note 1: This STC reflects the transfer of ENAC STC CS-091 from LEAT S.r.l. to VITROCISET S.p.A. This STC supersedes ENAC STC CS-091.</td>
<td>Doc. No. 100-0203-000-000 Master Drawing List dated 07.07.2003 and later approved revisions.</td>
<td>Included in the Supplemental Flight Manual No. 100-0203-800-000 dated 01.03.2003 and later approved revisions. This Certificate shall remain valid unless otherwise surrendered or revoked.</td>
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<td>JET AVIATION BUSINESS JETS AG</td>
<td>REV.1</td>
<td>Activation of the Runway Awareness and Advisory System (RAAS)</td>
<td>- Jet Aviation Declaration of Compliance ref BGL3343461 dated 03/02/2009 - Jet Aviation Master Data List No. BGL3343461-01, Revision IR dated 03/02/2009 - Jet Aviation AFMS No. BGL3343461-014-01, Revision IR dated 03/02/2009.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the aircraft and the installation of the product. The installation of this modification by thirds persons is sub...</td>
<td>25/02/2009</td>
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<td>GULFSTREAM AEROSPACE Ltd.</td>
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<td>GULFSTREAM GV-SP</td>
<td>Re-plate all seats &quot;Not for use during Taxi Takeoff &amp; Landing&quot;.</td>
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This modification shall introduce no adverse effect upon the airworthiness of the aircraft and the installation of the product. The installation of this modification by thirds persons is sub...
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<td>AEROSPATIALE ATR 42-500 AEROSPATIALE ATR 42-400 AEROSPATIALE ATR 42-300 AEROSPATIALE ATR 42-200</td>
<td>Installation of Honeywell KLN-90B System with dual CDr</td>
<td>See Scandinavian Avionics Design documents</td>
<td>Certification Documents 2051746 DOC issue 1 Engineering Order 2051746-EO Issue 2 Aeroplane Flight Manual Supplement 2051746-AM Issue 2 Continued Airworthiness 2051746-CA Issue 2 Maintenance Manual Supplement 2051746-MM Issue 30</td>
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<td>Dual FMS UNS-1lw System Installation</td>
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<td>A310 MODELS</td>
<td>EHS upgrade on Honeywell TRA-47486-01127-1402This STC upgrades existing transponder system by installing Mode SEnhance surveillance (EHS)</td>
<td>Master Data List ref. AED2008-10M_MDL Revision 4, dated 19 May 2009 (or any later approved revision) AFM Supplement, ref. AED2008-10M_MM 03 dated 20 December 2008</td>
<td>None.</td>
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<td>A320-200 MODELS</td>
<td>* Modification Summary, ref. doc. MOD-0648, Rev. A, dated 8th May 2009 Modification Compliance List, ref. doc. MCL-0548 Rev. A, dated 11th May 2009</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by third persons...</td>
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<td>FALCON 900EX</td>
<td>Activation of the Runway Awareness and Advisory System (RAAS) i.a.w. JetAviation Master Data List BF9EX34461-001 Revision B dated 09July 2009 Jet Aviation Aeroplane Flight Manual Supplement BF9EX34461-201-01Rev1 Revis on IR Jet Aviation Engineering Order (installation) BF9EX34461-344600Rev1 on IRs later EASA Approved revisions of these documents</td>
<td>* Jet Aviation Master Data List BF9EX34461-001 Revision B dated 09July 2009 Jet Aviation Aeroplane Flight Manual Supplement BF9EX34461-201-01Revision on IR Jet Aviation Engineering Order (installation) BF9EX34461-344600Revis on IRs later EASA Approved revisions of these documents</td>
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<td>GULFSTREAM III</td>
<td>Activation of the Runway Awareness and Advisory System (RAAS)EGPWC upgrade to software level 228-228 and activation of the RAAS</td>
<td>Installation in accordance with Jet Aviation Engineering Order BGLF334461 and Master Drawing List Doc No. BGLF334461-60 Rev 1 or later approved revisions. Operation in accordance with Jet Aviation Airplane Flight Manual Supplement of BGFU229441-201-01 Rev IR or later approved revisions.</td>
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<td>Dual SN100 EHSI Installation (Sandel)</td>
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<td>AVIONICS INTEGRATION &amp; ENGINEERING CO. RPORATION AG (AIEC)</td>
<td>CESSNA 206 SERIES</td>
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<td>Dual GNS 430W with TASS 600 System/installed two GARMIN GNS430W with external 121.5kHz Frequency selection and an Avidyne TASS 600 system</td>
<td>Master Data List. Doc Number 808-049AFMS Dual GNS430W with TASS600. Doc Number 808-07 Instruction for continued Airworthiness: Doc Number 808-065</td>
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<td>Aspen EFD-1000 Primary Flight Display must be installed in the aircraft.</td>
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<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td>PIPER PA-34</td>
<td>Pneumatic deicer removal.</td>
<td>SEG A application document no. A-03-09 Deice, DA edition 20 dated 20 March 2009 SEG A Flight Manual Supplement no. A-03-09-Deice edition 0 dated 20 March 2009 SEG A modification file no. A-03-09-Deice edition 0 dated 20 March 2009</td>
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<td>CESSNA DUESBeldorf CATION SNICERIE E CENTER GmbH</td>
<td>CESSNA 560</td>
<td>Installation of dual Honeywell GNS-XLS PRNAV FMS</td>
<td>- Engineering Order 2009-2 Revision D from 12 May 2006-AMF Supplement 2009-2.8 Revision B from 12 May 2009-Maintenance Manual Supplement 2009-2-10 Revision A from 3 March 2009-Declaration of Compliance of project 2006-2 from 20 May 2006 or later approved revisions</td>
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<td>REV 2</td>
<td>BOEING 737-700 IGW</td>
<td>CTT Zonal dryer activationReplacement of Zonal Dryer related to EASA AD 2006-0140-E. The above reference EASA AD required the de-activation of Zonal Drying Systems installed via Swiss FOCA STC Z 25-20-77 due to the potential for combustion of contaminants in the dryer unit. A new dryer unit has been...</td>
<td>CTT SB 2001-006-21-01-AD Initial EASA AD 2006-0140-E or later approved revisions Master Data List BBLF32071-001 Rev. C dated 15.01.2009 or later approved revision</td>
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<td>AIR ALLIANCE GmbH</td>
<td>B300, B300C</td>
<td>Installation of Honeywell Mark VI EGPWS.</td>
<td>- According to AirAlliance Master Document List Doc No. 2009_006_A242E AA Form 2, Rev. 0, or later approved Revisions and AirAlliance Installation Instructions 2009_006_A242E 500, Rev. 0, or later Revisions. To be operated in accordance with AirAlliance AFM-S Document No. 2009_006_A242E 500...</td>
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<td>EASA.A.S.03852</td>
<td>AIR ALLIANCE GmbH</td>
<td>PIPER PA-31T2, PIPER PA-31T1, PA-31T (CHEYENNE/CHEYENNE II)</td>
<td>Installation/Exchange of Avionics acc. AirAlliance MDL 2008_059_A8EA in latest approved Revision</td>
<td>According to AirAlliance Master Document List Doc. No. 2008_059_A8EA Rev. 0, or later approved Revisions and AirAlliance Engineering Order 2008_059_A8EA-050, Rev. 0, or later Revisions</td>
<td>STC holder</td>
<td>Rev.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>01/04/2009</td>
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<td>1015210</td>
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<td>EASA.A.S.03853</td>
<td>AIR ALLIANCE GmbH</td>
<td>PA-31T, PA-31T1, PA-31T2</td>
<td>Installation of Avionyx EX-500, TWX 670 and MLX 770 acc. AirAlliance MDL2009_003_A8EA in latest approved Revision</td>
<td>According to AirAlliance Master Document List Doc. No. 2009_003_A8EA Rev. 0, or later approved Revisions</td>
<td>STC holder</td>
<td>Rev.</td>
<td>none</td>
<td>20/05/2009</td>
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<td>1043081</td>
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<td>EASA.A.S.03858</td>
<td>SCANDINAVIAN AVIONICS DESIGN Aps</td>
<td>BEECH 400, 400A</td>
<td>Installation of various Entertainment Systems acc.: Engineering OrderDoc.: 2051195-EO Compliance Record Doc.: 2051195-COR</td>
<td>CS-25</td>
<td>STC holder</td>
<td>Rev.</td>
<td>28/07/2009</td>
<td>Active</td>
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<td>EASA.A.S.03859</td>
<td>H4 AEROSPACE (UK) Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>AIRBUS A320-232</td>
<td>GUEST/STAFF SEATING AREA RECONFIGURATION &amp; MISC CABIN ITEMS/AVIONS A320-232 AIRCRAFT as defined in HAACP057 Model: Airbus A320-232 MSN 2403, A6-DLM</td>
<td>STC holder</td>
<td>Rev.</td>
<td>Approval valid only for MSN 2403. The use of the aircraft is limited to non-commercial operation only. For other Limitations refer to Flight Manual Supplement H4AFMS019.</td>
<td>04/06/2009</td>
<td>Active</td>
<td>1015209</td>
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<td>EASA.A.S.03862</td>
<td>AMES AEROSPACE &amp; MECHANICAL ENGINEERING Services GbR</td>
<td>FALCON 2000</td>
<td>Falcon 2000 MIND209 Interior Refurbishment.</td>
<td>- EO_M08800101_MBO_14 04/2009-ICA_M08800101_MBO_1 4042009-EB_M08800101_MBO_14 04/2009</td>
<td>STC holder</td>
<td>Rev.</td>
<td>1. Limited to A/C Falcon 2000, MSN 209. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The...</td>
<td>11/06/2009</td>
<td>Active</td>
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<td>EASA.A.S.03870</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING (AVI) RATION AG (AIEC)</td>
<td>TASI 620 System Installation by installing an Avionyx TASI 620 TrafficAdvisory System</td>
<td>FAR 23</td>
<td>- Master Data List. Doc. No. 823-5-06, Rev. A or later approved revision - AFMS, Doc. No. 818-7-01, Rev. A or later approved revision - ICA, Doc. No. 818-0-06, Rev. 0 or later approved revision</td>
<td>STC holder</td>
<td>Rev.</td>
<td>None</td>
<td>03/07/2009</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>EASA.A.S.03886</td>
<td>JET AVIATION AG, BASEL</td>
<td></td>
<td></td>
<td>FALCON 900</td>
<td>Activation of the Runway Awareness and Advisory System (RAAS) i.a.w. Jet Aviation Master Data List BF90034461-001</td>
<td></td>
<td>-Jet Aviation Master Data List BF90034461-001 Revision A dated 19 May 2009 -Jet Aviation Aeroplane Flight Manual Supplement BF90034461-201-01 Revision IR-Jet Aviation Engineering Order (installation) BF90034461-34460 Revision 100 later EASA Approved revisions of these documents</td>
<td>Limitations and Conditions: 1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) The installation of this modification by third party...</td>
<td>22/05/2009</td>
<td>Active</td>
<td>10015304</td>
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<tr>
<td>EASA.A.S.03897</td>
<td>RUAG AEROSPACE SERVICES GmbH</td>
<td></td>
<td></td>
<td>EMBRAER EMB-135BJ</td>
<td>Stone Flooring / Replacement of existing floor covering by a stone floor covering in the forward entrance area and in the rear lavatory. This modification will not affect the structural flooring, the stone floor covering is only a decorative component.</td>
<td></td>
<td>Master Document List 289B-MDL Rev. A dtd. 25 May 2009</td>
<td>EASA EWS Generic Special Condition not applicable for this STC</td>
<td>26/05/2009</td>
<td>Active</td>
<td>10015306</td>
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<tr>
<td>EASA.A.S.03899</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td></td>
<td></td>
<td>BEECH E90 BEECH C90 BEECH B90</td>
<td>Be 'Mode S / Flight ID' capable Collins TDR-94(D) transponders</td>
<td></td>
<td>1) The installation will be made in accordance with Business and Commuter Aircraft document BCA108-05, Revision 0, as of 14.08.2008 or later EASA approved revision. 2) The Aircraft Flight Manual Supplement applicable to this modification is BCA108-03 document, Revision 0, as of 14.08.106B or later...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/04/2009</td>
<td>Active</td>
<td>10015307</td>
</tr>
<tr>
<td>EASA.A.S.03900</td>
<td>BUSINESS AND COMMUTER AIRCRAFT</td>
<td></td>
<td></td>
<td>BEECH F90</td>
<td>Be 'Mode S / Flight ID' capable Collins TDR-94(D) transponders.</td>
<td></td>
<td>1) The installation will be made in accordance with Business and Commuter Aircraft document BCA108-05, Revision 0, as of 14.08.2008 or later EASA approved revision. 2) The Aircraft Flight Manual Supplement applicable to this modification is BCA108-03 document, Revision 0, as of 14.08.106B or later...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>27/04/2009</td>
<td>Active</td>
<td>10015308</td>
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<tr>
<td>EASA.A.S.03911</td>
<td>CESSNA DUESSELDORF CITATIONSERVIC E CENTER GmbH</td>
<td></td>
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<td>BEECH 1900</td>
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<td>15/07/2009</td>
<td>Active</td>
<td>10228548</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>EASA.A.S.C.01065</td>
<td>DASSAULT FALCON SERVICE</td>
<td>MF 20 GF S/N 397</td>
<td>Installation of a probe dispenser system and of an UHF antenna</td>
<td>1 - Applicable only to MF 20 S/N 397; this STC is approved only for the configuration as defined in ModSheet DFS 2490/3Hx SIC is approved only for the product configuration as defined in the approved data referred to in the paragraph “Description”</td>
<td>17/06/2005</td>
<td>Active</td>
<td>10043154</td>
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<td>EASA.A.S.C.00066</td>
<td>DASSAULT FALCON SERVICE</td>
<td>EASA A S 00723</td>
<td>Installation of a probe dispenser system and of an UHF antenna</td>
<td>1 - Applicable only to MF 20 S/N 397, this SIC is approved only for the product configuration as defined in the approved data referred to in the paragraph “Description”</td>
<td>17/06/2005</td>
<td>Active</td>
<td>10043155</td>
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<td>EASA.BA.S.01001</td>
<td>CAMERON BALLOONS LIMITED</td>
<td>CAA UK RESPONSIBILITY</td>
<td>Introduction of Liquid Pilot Lights.</td>
<td>None.</td>
<td>30/04/2007</td>
<td>Active</td>
<td>10015308</td>
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<tr>
<td>EASA.E.S.01000</td>
<td>THIELENT AIRCRAFT ENGINES GmbH</td>
<td></td>
<td>Installation of the FADEC G8 in accordance with Installation Manual IM-80-01, Conversion Instruction No. IM-80-01, - Conversion Instruction for operation, maintenance and inspection; -Conversion Instruction for operation, maintenance and inspection</td>
<td>Engines having been converted via this STC may be identified as IO-580-B1A-ETE: The data in the TCDS EASA.IM.E.027 remains valid, except the following:Weight of the engine remains valid, except the following:Weight of the engine equipped with FADEC:</td>
<td>03/06/2007</td>
<td>Active</td>
<td>10015310</td>
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<tr>
<td>EASA.IM.A.02449</td>
<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>FAA STC ST10578/SC-D</td>
<td>EASA Validation of FAA approved STC ST10578/SC-D: Aircraft Compliance for Falcon 900EX serial number aircraft 176 in accordance with Data List FS-EX-0701-D179 Revision A dated 29th May 2007 or later revisions</td>
<td>1. Prior to installation of this modification the installationmustdetermine that the interrelationship between this modification and anything previously installed modification introduce no adverse affect upon the airworthiness of the product. The installation of this modification by third persons</td>
<td>09/07/2009</td>
<td>Active</td>
<td>10043206</td>
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<tr>
<td>EASA.IM.A.C.01293</td>
<td>PIPER AIRCRAFT INC.</td>
<td>PA-34-220T (SENECA V)</td>
<td>Installation of the Avistyle Entegra System Option (PFD/MFD) onto the PA-34-220T (Seneca V).</td>
<td>See EASA IM.A.090 (TCDS)</td>
<td>26/06/2006</td>
<td>Active</td>
<td>10042554</td>
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<tr>
<td>EASA.IM.A.S.0013</td>
<td>JET CORP</td>
<td>FAA STC ST00704WI</td>
<td>MYSTERE FALCON 900</td>
<td>Installation of an Allied Signal Enhanced Ground Proximity Warning System (EGPWS) with windshear detection and alerting in accordance with JetCorp STC Drawing List 13-327L,001( )&lt; &gt; Revision 6, dated 20NOctober 2000 (or later EASA approved revision).</td>
<td>1. The installation of the change, subject of this STC, must be associated with the installation of a change modifying the color ground of the TERRAIN INHIBIT switch. The accepted colors are amber, white, orange. 2. EASA approved Airplane Flight Manual Supplement JetCorp N° 13-327B001Revision B( )&lt; &gt; dated...</td>
<td>24/02/2004</td>
<td>Active</td>
<td>10062941</td>
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<td>EASA.IM.A.S.00144</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA3333WE</td>
<td>B727-100/-200 SERIES</td>
<td>1 The approval of this change applies basically to the Boeing 727-100 -200 series only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configuration shall be determined...</td>
<td>1. The installation of the change, subject of this STC, must be associated with the installation of a change modifying the color ground of the TERRAIN INHIBIT switch. The accepted colors are amber, white, orange. 2. EASA approved Airplane Flight Manual Supplement JetCorp N° 13-327B001Revision B( )&lt; &gt; dated...</td>
<td>07/10/2004</td>
<td>Active</td>
<td>10061359</td>
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<tr>
<td>EASA.IM.A.S.00145</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA3334WE</td>
<td>737-400, 737-500, 737-200, 737-300</td>
<td>Installation of Stretch Carrier in accordance with AMEDIC Master Drawing List Ref: AM-2001, Revision 3, dated August 13, 1998 (or later EASA approved Revisions).</td>
<td>1. The approval of this change applies basically to the Boeing 737-100, -200, -300, -400 and -500 series only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configuration shall be determined...</td>
<td>07/10/2004</td>
<td>Active</td>
<td>10061360</td>
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<td>EASA.IM.A.S.00146</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA3335WE</td>
<td>2. DC-10-40</td>
<td>Installation of Stretch Carrier in accordance with AMEDIC Master Drawing List Ref: AM-2001, Revision 3, dated January 28, 1977 (or later EASA approved Revisions).</td>
<td>1. The approval of this change applies basically to the McDonnellDouglas DC-10-40 series only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configuration shall be determined...</td>
<td>07/10/2004</td>
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<td>10061369</td>
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<td>EASA.IM.A.S.00147</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA3511WE</td>
<td>747SR (US A20WE) 747-300, 747-400, 747SP 747-100, 747-200B, 747-200C</td>
<td>Installation of Stretch Carrier in accordance with AMEDIC Master Drawing List Ref: AM-2001, Revision 4, dated February 28, 1978 (or later EASA approved Revisions).</td>
<td>1. The approval of this change applies basically to the Boeing 747-100, -200, -300, -400 and -500 series only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configuration shall be determined...</td>
<td>07/10/2004</td>
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<td>EASA.IM.A.S.00148</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA3499NM</td>
<td>757-200, 757-300</td>
<td>Installation of Stretch Carrier in accordance with AMEDIC Master Drawing List Ref: AM-2001, Revision 4, dated March 11, 1987 (or later EASA approved Revisions).</td>
<td>1. The approval of this change applies basically to the Boeing 757-200 series only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configuration shall be determined...</td>
<td>07/10/2004</td>
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<td>10061363</td>
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<td>EASA.IM.A.S.00149</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA2023NM</td>
<td>B767-200, -300, -400ER</td>
<td>Installation of Stretch Carrier in accordance with AMEDIC Master Drawing List Ref: AM-2001, Revision 4, October 27, 1983 (or later EASA approved Revisions).</td>
<td>1. The approval of this change applies basically to the Boeing 767-200 series only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configuration shall be determined...</td>
<td>07/10/2004</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>EASA.IM.A.S.0015</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>Installation of Rosemount Aerospace Inc. (RMTAERO) Angle of Attack sensor model 0861 HB, in accordance with Amtech Aeronautics Limited Modification Data Summary MDS317601 (RMTAERO) document number D0000010, Revision 1, dated 27 July 2001, or later EASA approved revision.</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, the STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this certificate is requested under Part 21.A.116, the STC will be reassigned. 3. This STC is approved only for the product.</td>
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<td>26/06/2004</td>
<td>Active</td>
<td>10013575</td>
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<td>EASA.IM.A.S.00150</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA3653WE DC-8-60 SERIES</td>
<td>1. The approval of this change applies basically to the McDonnellDouglas DC-8-60 passenger versions only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph.</td>
<td></td>
<td></td>
<td>07/10/2004</td>
<td>Active</td>
<td>10061370</td>
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<td>EASA.IM.A.S.00162</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST3106LA-D</td>
<td>BOMBARDIER BD-100-1A10 (CHALLENGER 350) The modification consists in the installation of an Executive Interior in accordance with Master Drawing List HCC2001355001, Rev. E dated July 16, 2004 (or later FAA approved revision). Validation of FAA STC ST3106LA-D.</td>
<td></td>
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<td>07/10/2004</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>REV. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>EASA.IM.A.S.00166</td>
<td>AIRCRAFT SYSTEMS AND MANUFACTURING, Inc.</td>
<td>FAA STC ST0687 SC</td>
<td>BOEING 727</td>
<td>Installation of RVSM compliant equipment (dual DADS) FAA STC N° ST0687 SC</td>
<td>1. AirWorks Airplane Flight Manual Supplement (AFM-S) Document NoAFMS23010006, Revision Original. FAA approved on February 5, 2003, or later FAA approved revision, is required as part of the modification. This STC is approved only for the product configuration as defined in the approved design data...</td>
<td>07/10/2004</td>
<td>Active</td>
<td>10015313</td>
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<tr>
<td>EASA.IM.A.S.00199</td>
<td>AIRWORKS</td>
<td>FAA STC ST01432LA</td>
<td>A321-211,-211, A321-111,-112,-113, A320-231,-232,-233, A320-111,-211,-212,-214, A319-114,-131,-132, A319-111,-112,-113</td>
<td>Installation of AirWorks Cockpit Door Surveillance Systems in accordance with AirWorks Index List Document No IL23001, Revision NC, dated February 5, 2003 or later FAA approved Revision. This STC is a validation of the FAA-approved STC No ST01432LA.</td>
<td>1. AirWorks Airplane Flight Manual Supplement (AFM-S) Document NoAFMS23010006, Revision Original. FAA approved on February 5, 2003, or later FAA approved revision, is required as part of the modification. This STC is approved only for the product configuration as defined in the approved design data...</td>
<td>14/10/2004</td>
<td>Active</td>
<td>10043111</td>
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<tr>
<td>EASA.IM.A.S.0022</td>
<td>MACHINING PROGRAMMING MANUFACTURING, Inc.</td>
<td>FAA STC</td>
<td>B7-100-1410</td>
<td>The modification consists in the installation of folding step ladder to provide access to the tailcone equipment compartment in accordance with MPM Inc. Aft Ladder Installation Drawing LD001 dated 4th September, 2003 (or later EASA approved revision)</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by installer.</td>
<td>08/07/2004</td>
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<td>EASA.IM.A.S.00228</td>
<td>FRAKES AVIATION</td>
<td>FAA STC</td>
<td>B300C B300 B200C B200 300LW 300 200 200C 1900C 1900</td>
<td>Installation of engine exhaust stub in accordance with Frakes Aviaion/Installation drawing number FA5726, dated September 6, 1985, for the 200series, and drawing FA5601 dated June 19, 1995 for the 300 and 1900 series, or later EASA approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by installer.</td>
<td>28/10/2004</td>
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<td>EASA.IM.A.S.00244</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC</td>
<td>SA53215SW-D</td>
<td>BOEING 767-300ER</td>
<td>Reconfiguration of passenger cabin interior from 204 passengers to 250 passengers in accordance with DER Services Inc. Master Document List (MDL) Report No. MDL-A2004-0261 dated 15 June 2004, or later EASA approved revisions.</td>
<td>1. Subject to compliance with the provisions of Part 21Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this Certificate is requested under Part 21A, it shall be released. This approval is limited...</td>
<td>14/10/2004</td>
<td>Active</td>
<td>10015317</td>
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## List of EASA Supplemental Type Certificates

<table>
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<tr>
<th>EASA STC No.</th>
<th>STC Holder</th>
<th>Rev.</th>
<th>Foreign/NAA STC No.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<th>System Record</th>
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<tr>
<td>EASA.IM.A.S.00260</td>
<td>EMPIRE AIRLINES</td>
<td>FAA STC ST01448SE</td>
<td>ATR 42/72</td>
<td>Installation of B/E Aerospace Sweep-On Fullface Crew OxygenMask System on the flight deck of ATR 42/72 series airplanes in accordance with Empia Airlines Document Number 3510, Revision Original dated May 25, 2004, or later FAA-approved revision. This STC is a validation of the FAA-approved STC No...</td>
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<td>EASA.IM.A.S.00275</td>
<td>HAWKER BEECHCRAFT LTD.</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BEECH 400A</td>
<td>Installation of a Honeywell Enhanced Ground Proximity Warning System (EGPWS) Mark VII in accordance with Elliot Aviation Master Data List 2018-571DX, Revision 4, dated 6 November2003 or later EASA approved revisions: This STC is issued asease of the validation of the Elliot Aviation Technical Pro...</td>
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<td>20/10/2004</td>
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Note: The table contains information on various Supplemental Type Certificates issued by the European Union Aviation Safety Agency (EASA), including the STC number, STC holder, revision level, foreign/NAA STC number, model, description, certification basis, associated technical documentation, limitations/conditions, date of issue, status, and system record. The table also includes specific details such as FAA STC numbers, model designations, and dates of approval.
### List of EASA Supplemental Type Certificates

<table>
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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
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<th>Foreign/NAACert ref.</th>
<th>Model</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tr>
<td>EASA.IM.A.S.00289</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA0264AC-D</td>
<td>CESSNA F182-Q</td>
<td>Installation of S-TEC System 30 Two Axis Automatic Flight Guidance System, Model ST-718/30, according to Bulletin No. 818, dated July 30, 1997 and Master Drawing List No. 921016, dated July 30, 1997 or later FAA approved revisions of the above data (28 Volt System). This STC is issued as result of...</td>
<td>1. EASA Approved Pilot’s Operating Handbook and/or Airplane Flight Manual Supplement, dated October 21, 2004 is required for Reims-Cessna Model F182Q for S-TEC System 30 or later EASA approved revisions of the above supplement.2. Compatibility of this modification with other previously app...</td>
<td>28/10/2004</td>
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<td>EASA.IM.A.S.00313</td>
<td>G&amp;D AERO PRODUCTS INC</td>
<td>FAA STC SA759NM</td>
<td>V35, V36A, V36B,36, A36, A38TC, F33C, H35, K35, N35, P35, S35, B36TC, 35-C33A, E33, E33A, F33A, 35-33, 35-A33, 35-B33, 35-C33,</td>
<td>Installation of acrylic windowliners in accordance with FAA approved Report No1., revision B, dated March 15, 1995 or later EASA-approved revision.</td>
<td>1. This installation should not be incorporated in any aircraft unless it is determined that the interrelationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of the aircraft.2. This STC is issued as result...</td>
<td>16/11/2004</td>
<td>Active</td>
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<td>EASA.IM.A.S.00322</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST00151BO</td>
<td>LEARJET 55</td>
<td>Installation of Honeywell CAS-67A TCAS II Software Change 7.0.</td>
<td>1. FAA STC ST00151BO:2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;: Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>09/11/2004</td>
<td>Active</td>
<td>10015339</td>
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<td>EASA.IM.A.S.00331</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST00389BO</td>
<td>60</td>
<td>Installation of an ICG Indium Satcom System FAA STC ST01569LA-D Inconformance with Learjet Master Document List LC33156000, Revision F, FAAApproved May, 27th 2004, or later FAA Approved revisions.</td>
<td>1. FAA Approved AFM Supplement L-FM23155000 Revision NO, dated May 20th, 2004 or later FAA Approved revision.2. Subject to compliance with the provisions of Part 21 Subpart E, thisSTC and associated data shall remain valid until surrendered, withdrawn otherwise terminated.3. If transfer of this Certi...</td>
<td>25/11/2004</td>
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<td>EASA.IM.A.S.00345</td>
<td>SIMMONDS PRECISION PRODUCTS, Inc</td>
<td>FAA STC ST00838BO</td>
<td>BOEING 727-200 SERIES</td>
<td>Installation of either the Goodrich Corporation Fire Detection and Suppression system, or the Fire Detection only system in accordance with Goodrich Corporation Master Drawing List T3038-MDL-0101, Revision A, dated October 31, 2003 or later FAA-approved revision; as defined in FAA STC ST00838BO, iss...</td>
<td>1. The suppression feature of Fire Detection and Suppression system has been certified for a maximum of 60 minutes from the activation of the suppression agent. Aircraft diversion time, including landing, should not exceed 60 minutes.2. This STC is approved only for the product configuration...</td>
<td>30/11/2004</td>
<td>Active</td>
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<tr>
<td>EASA.IM.A.S.00350</td>
<td>HAWKER BEECHCRAFT Ltd</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>BEECH 1900C, ARLINER</td>
<td>Defines all cargo interior using web nets. The all cargo interior is defined by the top drawing of Raytheon AircraftServices No114-0100, no change, dated April 17, 1996, with an &quot;FAA Approved&quot; stamp of April 17, 1996 or later FAA-approved revision.</td>
<td>1. The Airplane Flight Manual Supplement No RAS 114-990021 is required equipment when this STC is installed.2. Compatibility of this design change with previously approved modifications must be determined by the installer.3. This STC is issued as a result of the validation of the Raytheon A...</td>
<td>03/12/2004</td>
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<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>RAYTHEON AEROSPACE SERVICES, INC.</td>
<td>EASA.IM.A.S.00352</td>
<td>FAA SAU0479WI BEECH 1900C Installation of a cargo web &quot;Z&quot; net in the forward fuselage and robotics chair. The &quot;Z&quot; net and robotics chair cannot both be installed in the airplane at the same time. Installation of the two configurations per Raytheon Aerospace Aircraft System drawing 1900U1100, n/c, dated September 20, 1994, and...</td>
<td>FAA</td>
<td>1. The FAA approved Aircraft Flight Manual Supplement p/n 1900G001 dated November 6, 1996, or later FAA-approved revision is required equipment, when this STC is installed. 2. Compatibility of this design change with previously approved modifications must be determined by the installer. 3. This STC is...</td>
<td>07/12/2004</td>
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<td>10015335</td>
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<td>BOMBARDIER INC.</td>
<td>EASA.IM.A.S.00358</td>
<td>TOCCA LT STC C LSA04-213/D CL-600-2B16 Installation of Sideview Camera System Transport Canada Limited STC C-LSA04-213/D in accordance with Aero Consulting Services Modification Data Summary Notice E091000 Revision n/c or later Transport Canada Approved Revision.</td>
<td>FAA</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. Transfer of this Certificate is requested under Part 21A.116, the STC will be reissued. 3. This approval is limited to...</td>
<td>07/12/2004</td>
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<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>EASA.IM.A.S.00360</td>
<td>FAA STC ST01449CH BOEING 757-200 Dual GPS Antenna and Cox Cable (FAA STC ST01449CH) as per Master Data List ECS-200555 revision B dated June 11, 2003 or later FAA-approved revision.</td>
<td>FAA</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. Transfer of this Certificate is requested under Part 21A.116, the STC will be reissued. 3. This STC is approved only f...</td>
<td>07/12/2004</td>
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<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>EASA.IM.A.S.00361</td>
<td>FAA STC ST01839CH BOEING 757-200 Honeywell MK V Enhanced Ground Proximity Warning System (FAA STC ST01839CH) as per Master Data List ECS-201736 revision H dated 25 May 2004 or later FAA-approved revision.</td>
<td>FAA</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. Transfer of this Certificate is requested under Part 21A.116, the STC will be reissued. 3. This STC is approved only f...</td>
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<td>AVIATION FABRICATORS, Inc.</td>
<td>EASA.IM.A.S.00389</td>
<td>REV. 2 FAA SA00835WI 2.A200C, A200CT, B200T, B200CT 2. 200C, 2000C, 200CT, A250 1. B200, B200C Installation of all Jump Seat Kits.</td>
<td>FAA</td>
<td>1. Installation per the STC Drawing Lists defined in the MIL is limited to aircraft with Beech factory installed oxygen system for the jump seat locations. Aircraft configured for more than 14 total occupants must be equipped with a 49 cubic foot or greater oxygen system.</td>
<td>17/04/2008</td>
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<td>STC Holder</td>
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<td>Description</td>
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<td>Limitations/Conditions</td>
<td>Associated Technical Documentation</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>ROCKWELL COLLINS, Inc.d/b/a AIR TRANSPORT SYSTEMS</td>
<td>FAA STC ST013220W-D,STC EASA</td>
<td>Modification of an existing FAA-approved Collins TPR-720, TPR-900 or TPR-901 Mode S transponder installation in Part 25 aircraft.</td>
<td>- FAA STC No ST013220W-D, Collins Drawing Index as listed on AML No ST013220W-D, revision original, or later FAA-approved revisions.</td>
<td>1. Prior installation and approval of a Collins TPR-720, TPR-900 or TPR-901 Mode S transponder is a prerequisite to this STC. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft.</td>
<td>29/03/2005</td>
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<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA A19WE</td>
<td>Installation of a Honeywell Enhanced Ground Proximity Warning System in accordance with Electronic Cable Specialists Master Data List ECS-201532, Revision B, dated March 17, 2003.</td>
<td>- FAA STC ST01798DC amended on 24/4/2004 (original STC for Boeing 737-300/400 was approved in December 2003).</td>
<td>1. FAA Supplement as listed in Appendix A of ECS Master Data List Rev. TEC201532 FAA Approved 23.11.2004. For configuration 13, GPS antenna installation must be accomplished prior to this modification in accordance with FAA STC ST094095C. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>14/12/2004</td>
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<td>AIM AVIATION RENTON, INC.</td>
<td>757-200</td>
<td>Installation of overhead Stowage Bin Doors.</td>
<td>- AIM Aviation Inc. Master Data List AIM-02-1060 Rev. A dated 06/12/2004 or later FAA-approved revision.</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, the STC and associated shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this STC is requested under Part 21 Subpart E, the STC will be updated as necessary. 3. This STC is approved for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>21/12/2004</td>
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<td>HONEYWELL INTERNATIONAL INC.</td>
<td>FAA STC ST051755E</td>
<td>Installation of the Enhanced Ground Proximity Warning System on a Boeing 747-200-200 aircraft.</td>
<td>Master Drawing, Enhanced Ground Proximity Warning System Boeing 747 Series Aircraft, Honeywell. document 966-459-094, rev. BC.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer. For Boeing 874-2001, in addition to the FAA-approved Aircraft.</td>
<td>29/01/2007</td>
<td>Active</td>
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<td>ROCKWELL COLLINS, Inc.d/b/a AIR TRANSPORT SYSTEMS</td>
<td>FAA STC ST0587W-D</td>
<td>Modification of Collins ACARS DLM-700B Management Unit (MU) System Installation to Collins DLM-900 MU System Installation.</td>
<td>- FAA STC ST0587W-D, Collins Drawing Index 963-4300-011, no revision, dated 29/2/2007 and Maintenance Manual Supplement (MMS) ACC-97-131, No revision, dated 10/17/98.</td>
<td>1. Collins ACARS DLM-700B Management Unit System Installation. EASA STC EASA.IM.A.S.0043, is required to be installed prior to or concurrent with using this STC. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”...</td>
<td>21/12/2004</td>
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<td>FAA STC No.</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
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<td>Status</td>
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<td>EASA.IMA.S.00446</td>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td>FAA ST02004AT</td>
<td>747-200, 747-300</td>
<td>12CAB provisions on Boeing 747-200 &amp; -300 Series Aircraft.</td>
<td>- FAA STC ST02004AT. - Avionics Support Group Inc Master Drawing List 24955-MDL, Revision IR dated June 22, 2004 other approved revision.</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, the STC and associated shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this STC is requested under Part 21A, the STC will be reissued. 3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>21/12/2004</td>
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<tr>
<td>EASA.IMA.S.00475</td>
<td>C &amp; D INTERIORS</td>
<td>REV. 1</td>
<td>BOEING 737-600</td>
<td>C&amp;D Interiors Stowage S1, Galley S1, Galley G2B, Galley G2B, Stowage S2, Forward lavatory, Passenger Seat Installation, Overhead Stowage Bin Installation, Sidewall Installation, PSU Installation, Passenger Seat Proximity Light installation.</td>
<td>- C&amp;D Interiors Index List A61 Revision B, dated August 28,2003 or later FAA-approved revision.</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, the STC and associated shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this STC is requested under Part 21A, the STC will be reissued. 3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>17/05/2006</td>
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<td>EASA.IMA.S.00489</td>
<td>RAM AIRCRAFT, LP</td>
<td>172, 172K, 172L, 172M, 172N, 172D, 172E, 172F, 172G, 172H</td>
<td>Installation of Lecture C-320-D2Q.G1,- D2QJ/-E2D engine models</td>
<td>FAA STC SA23755WR Rev. 8 dated 28 August 1987, AM, Drawing 17002 Rev. K dated 09.12.1986, RAM Drawing 17003 Rev. L dated 09.12.1986 or later FAA approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/01/2005</td>
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<td>EASA.IMA.S.00498</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>FAA STC ST011245E</td>
<td>BOEING 747-400 SERIES</td>
<td>Installation of the Flight Data Acquisition Management System</td>
<td>- Master Drawing List 796-0082-001, rev A, or later FAA approved revisions. - FAA STC ST011245E</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>10/01/2005</td>
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<tr>
<td>EASA.IMA.S.00503</td>
<td>ELLIOTT AERONAUTICS</td>
<td>REV. 1</td>
<td>BEECH B300C BEECH B300T BEECH B300CT BEECH B300B BEECH B300LW BEECH 300B BEECH 200T BEECH 300T BEECH 300LW</td>
<td>Installation of Dual TDR-4D Transponders with Flight ID B300/ B200C/B300/B300C Aircraft with Pro Line 21 Avionics, in accordance with Configuration 3 of Elliott Aviation Master Data List. No 2005-SRDL, Revision 2 or later approved revision.</td>
<td>- Airplane Flight Manual Supplement. Elliott Aviation Doc No. FS206-34 Rev. 0 or later EASA approved revision is required on board the aircraft.</td>
<td>1. Limited to Aircraft equipped with Raytheon Aircraft Company &quot;Colins Pro Line 21 Avionics Suite&quot;, &quot;software major change2. Limited to Configuration 3 of Elliott Aviation FAA STC SA011253CH. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>06/03/2006</td>
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<td>EASA.IMA.S.00502</td>
<td>AIRCRAFT SYSTEMS AND MANUFACTURIN G, INC</td>
<td>FAA STC ST006865C REV 1</td>
<td>DC-9-81 (MD-87) DC-9-81 (MD-83) DC-9-81 (MD-92) DC-9-81 (MD-81)</td>
<td>Installation of Universal UNIS Flight Management System with UL-601 Unlink Data Link System</td>
<td>- FAA STC ST006865C Rev 1-ASM Top Drawing List No 5542-0012-00 Rev T- AFM Supplements Doc No 5542-2160-00 Rev B and Doc No 5542-2161-01 or later approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>19/01/2005</td>
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<td>EASA.IMA.S.00531</td>
<td>HARTFORD PROPELLER INC.</td>
<td>FAA SA181CE</td>
<td>K36, M35, P35, 35-C36, E33, F33, 35-33, 35-A30, 35-B33,</td>
<td>Installation of Hartford PHC-L32YF-1RF/ F8464A-6R or BHC-L2YF-1RF</td>
<td>1. Limited to Aircraft equipped with Raytheon Aircraft Company &quot;Colins Pro Line 21 Avionics Suite&quot;, &quot;software major change2. Limited to Configuration 3 of Elliott Aviation FAA STC SA011253CH. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
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<td>EASA.IM.A.S.00543</td>
<td>SAFTGLO Inc.</td>
<td>FAA STC ST02248AT</td>
<td>A330-300 SERIES A330-200 SERIES</td>
<td>Installation of Saftglo Inc. photoluminescent floor proximity emergencesscape path marking system. (Validation of FAA approved STC ST02248AT)</td>
<td>- MDL L68-3 Rev. A dated Nov 15, 2000 or later FAA approved revision. - Flight Manual supplement L88-10 Rev I.R.</td>
<td>1. The approval of this modification is limited to the identified aircraft models. The limitations and conditions defined in the EASADatasheet are also applicable. A copy of the STC approval and the MDL must be enclosed in the files of each modified aircraft.</td>
<td>- Installation Instructions W3556005.</td>
<td>27/01/2005</td>
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<td>EASA.IM.A.S.00563</td>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA ST06486C</td>
<td>G250</td>
<td>Installation of side facing divan</td>
<td>- FAA STC ST01016SC-D - Index List GA222088001 Rev. D, dated 27/03/2003, or later FAA approved revision - FAA approved flight manual supplement GA22208M001, Rev. None, dated 27/03/2003 or later FAA approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>04/12/2008</td>
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<td>EASA.IM.A.S.00567</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>REV. 1 FAA STC ST10016SC-O</td>
<td>GULFSTREAM 200</td>
<td>Installation of BE Single (light weight) Seats Arrangement.</td>
<td>- FAA STC ST10016SC-D - Index List GA222088001 Rev. D, dated 27/03/2003, or later FAA approved revision - FAA approved flight manual supplement GA22208M001, Rev. None, dated 27/03/2003 or later FAA approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/02/2005</td>
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<td>EASA.IM.A.S.00568</td>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA STC ST01014SC-O</td>
<td>G250</td>
<td>Installation of Stainless Steel Lavatory Waste Ducting System</td>
<td>- FAA STC ST10016SC-D - Index List GA222088001 Rev. D, dated 27/03/2003, or later FAA approved revision - FAA approved flight manual supplement GA22208M001, Rev. None, dated 27/03/2003 or later FAA approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/02/2005</td>
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<td>EASA.IM.A.S.00569</td>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA ST06066SC-O</td>
<td>G230</td>
<td>Installation of Emergency Lighting Exit Signs and Floorpath Lighting System</td>
<td>- Installation Instructions W3556005.</td>
<td>- Installation Instructions W3556005.</td>
<td>09/02/2005</td>
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<td>EASA.IM.A.S.00598</td>
<td>WILCO, Inc.</td>
<td>FAA STC SA00741W</td>
<td>RAYTHEON 1900D RAYTHEON 1900C BEECH F90 BEECH E90 BEECH C90 BEECH C90A BEECH C90I BEECH E99 BEECH B60 BEECH B300 BEECH B200 T BEECH B200 C BEECH B100 BEECH A100 BEECH 99A BEECH 99B BEECH 65-400 BEECH 65-90 BEECH 300LW BEECH 300 BEECH 200T BEECH 200C BEECH 200 BEECH 100D BEECH 100</td>
<td>Installation of the Concorde RG-380 E/K Series VRSLA Batteries, as per FAA STC SA00741W.</td>
<td>- Installation Instructions W3556005.</td>
<td>- Installation Instructions W3556005.</td>
<td>02/03/2005</td>
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<td>ST01372SE</td>
<td>ISRAEL AEROSPACE INDUSTRIES, Ltd. (IAI)</td>
<td>REV. 1</td>
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<td></td>
<td>System, in accordance with Master Drawing List IA-103MDL</td>
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<td>1. Operation in accordance with AFM Supplement CS3790 Rev. B or later approved revision. AMM Supplement A65-10, older approved revision. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously...</td>
<td>22/10/2007</td>
<td>Active</td>
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<td>ST01595LA-D</td>
<td>AIRWORTHINESS RESOURCES CORPORATION</td>
<td>REV. 2</td>
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<td>Installation of a Service Panel Lighting System, in accordance with Master Drawing List HC33105000, Revision B dated April 23, 2004 or later FAA approved revision. This STC is the dated April 23, 2004 or later FAA approved revision.</td>
<td></td>
<td></td>
<td>1. Limitation on FAA STC in respect of concurrent installation of Thales IFE STC ST0162LA is not applicable. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations...</td>
<td>17/03/2005</td>
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<td>ST01603LA-D</td>
<td>LEARJET, Inc.</td>
<td>REV. 1</td>
<td></td>
<td>BOMBARDIER BD-100-1A10 (CHALLIE)</td>
<td></td>
<td>Installation of an Integrated Liquid Spray System (ILSS), relocation of the cockpit control panel for the Integrated Liquid Spray System (ILSS)</td>
<td></td>
<td></td>
<td>1. Modification of an aircraft by this supplemental TypeCertificate obligates the aircraft operator to include the maintenance information provided by Document Number HCS3105000-IFCA-1 in the operator's Aircraft Maintenance Manual and the operator's Aircraft Scheduled Maintenance Program.</td>
<td>28/07/2004</td>
<td>Active</td>
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<td>ST00515WI</td>
<td>LEARJET, Inc.</td>
<td>REV. 1</td>
<td></td>
<td>BOMBARDIER BD-100-1A10 (CHALLIE)</td>
<td></td>
<td>Installation of an Allied Signal CAS 67A Traffic Alert &amp; Collision Avoidance (TCAS II) (STC ST00515WI)</td>
<td></td>
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<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>04/04/2005</td>
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<td>ST01603LA-D</td>
<td>LEARJET, Inc.</td>
<td>REV. 1</td>
<td></td>
<td>BOMBARDIER BD-100-1A10 (CHALLIE)</td>
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<td>Installation of an Allied Signal CAS 67A Traffic Alert &amp; Collision Avoidance (TCAS II) (STC ST00515WI)</td>
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<td>1. Modification of an aircraft by this Supplemental TypeCertificate under -1 Configuration/Installation, obligates the aircraft operator to include the maintenance information provided by Document Number HCS3105000-IFCA-1 in the operator's Aircraft Maintenance Manual and the operator's Aircraft S...</td>
<td>28/07/2004</td>
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List of EASA Supplemental Type Certificates

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<th>STC No.</th>
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<td>EASA.IMA.0067</td>
<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>ST01253LA-D</td>
<td>Installation of Cockpit Writing Tables, in accordance with Master Drawing List HC31515000, Revision D, dated December 18, 2003 or later FAA-approved revision. This STC is the validation of the FAA approved STC No. ST01253LA-D.</td>
<td>FAA STC</td>
<td>Boulevard No. 811, dated 01.07.1997 and Master Document Li09111, dated 01.07.1997 or later approved revision (14 Volt System)/FAA-applicable POH and/or AFM-Supplement PN 819448, dated 07.07.1997 or later approved Revision for S-TEC System 20 on Piper PA-28-140, PA-28-150, PA-28-160, PA-28-180 S...</td>
<td>1. Modification of an aircraft by this Supplemental Type Certificate obligates the aircraft operator to include the maintenance information provided by Document Number HC31515000-IFCA in the operator's Aircraft Maintenance Manual and the operator's Aircraft Scheduled Maintenance Program.2. The...</td>
<td>28/07/2004</td>
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<td>10115300</td>
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<td>EASA.IMA.00676</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC</td>
<td>SA09254AC-D</td>
<td>Installation of S-Tec System 20/30 Single and Two Axis Automatic Flight Guidance Systems, Model ST-711-20/310.According FAA-STC SA09254AC-D</td>
<td>FAA STC</td>
<td>Bulletin No. 811, dated 01.07.1997 and Master Document Li09111, dated 01.07.1997 or later approved revision (14 Volt System)/FAA-applicable POH and/or AFM-Supplement PN 819448, dated 07.07.1997 or later approved Revision for S-TEC System 20 on Piper PA-28-140, PA-28-150, PA-28-160, PA-28-180 S...</td>
<td>1. Limitation according FAA STC SA09254AC-D. Compatibility of this modification with other previously approved modifications must be determined by the installer. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description...</td>
<td>11/09/2008</td>
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<td>EASA.IMA.00677</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC</td>
<td>ST01523SE</td>
<td>Installation of a Honeywell RESCU 460AF Emergency Locator Transmitter (ELT) System in accordance with Master Drawing List HC31515000, Revision D, dated December 19, 2003 or later FAA-approved revision. This STC is the validation of the FAA approved STC No. ST01523SE-D.</td>
<td>FAA STC</td>
<td>FAA-approved AFM supplement, document No. 14526015, dated 29 March 2005, or later FAA-approved revision is a required part of this STC.</td>
<td>1. The approval holder shall fulfill the obligations of Part 21, Paragraph 21.11.1. Configuration 1- ThisSTC must be installed concurrently with Jamco America ST01455SE- EJM Aerospace Services ST00730AT and Boeing Service Bulletin 747-25-3306, Revision 12 and FAA approved AFM supplement, document No. 14526015, dated 29 March 2005, or later FAA-approved revision is a required part of this STC.</td>
<td>11/10/2005</td>
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<td>EASA.IMA.0068</td>
<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>ST01254LA-D</td>
<td>Installation of a Honeywell RESCU 460AF Emergency Locator Transmitter (ELT) System in accordance with Master Drawing List HC31515000, Revision D, dated December 19, 2003 or later FAA-approved revision. This STC is the validation of the FAA approved STC No. ST01254LA-D.</td>
<td>FAA STC</td>
<td>FAA-approved AFM Supplement, document No. 14526015, dated 29 March 2005, or later FAA-approved revision is a required part of this STC.</td>
<td>1. The approval holder shall fulfill the obligations of Part 21, Paragraph 21.11.1. Modification of an aircraft by this Supplemental Type Certificate obligates the aircraft operator to include the maintenance information provided by this document in the operator's Aircraft Scheduled Maintenance Program...</td>
<td>28/07/2004</td>
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<td>EASA.IMA.00699</td>
<td>ARMSTRONG AEROSPACE Inc., ASTRONICS ARMSTRONG AEROSPACE</td>
<td>FAA STC</td>
<td>ST00687</td>
<td>Installation of a Honeywell RESCU 460AF Emergency Locator Transmitter (ELT) System in accordance with Armstrong Aerospace Master Data List Doc. No DC103-3016-00, revision B, dated December 9, 2003, or later FAA-approved revision/FAA-approved AFM Supplement, Armstrong Aerospace Doc. No DC303-3016-00, revision 1...</td>
<td>FAA STC</td>
<td>Armstrong Aerospace Master Data List Doc. No DC103-3016-00, revision B, dated December 9, 2003, or later FAA-approved revision/FAA-approved AFM Supplement, Armstrong Aerospace Doc. No DC303-3016-00, revision 1...</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.IM.A.S.00750</td>
<td>NORTHROP GRUMMAN GUIDANCE ANDELECTRONIC SYSTEMS COMPANY, INC.</td>
<td>REV 2</td>
<td>FAA STD ST01146A</td>
<td>AIRBUS A300 MODEL B4-203, B4-103, AIRBUS A300 MODEL B2-203</td>
<td>Installation of Litton Heading and Attitude System (LHAS LTR-97). This STC is the result of the FAA STC No. ST0149A validation.</td>
<td>Top installation drawing List. Document No. 454170-14 Revision C, dated 27 June 2001.</td>
<td>1. This STC is limited to the installation of three LHAS LTR-97 (Triple configuration). 2. The Airplane Flight Manual Supplement, Document No. 434599-EASA dated 29 May 2005 is required as part of this installation. This STC is approved only for the product configuration as defined in the approved STC.</td>
<td>31/07/2006</td>
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<td>EASA.IM.A.S.00762</td>
<td>HAWKER BEECHCRAFT SERVICES</td>
<td>REV 1</td>
<td>FAA STC SA0560DE</td>
<td>BEECH 1900C, 1900D</td>
<td>Installation of a L3/ACSS TCAS-2000</td>
<td>Genesis 3 Engineering Master Drawing List, Document No. GEN040311-002Revision A, dated September 2, 2004 or later EASA Approved revisions.</td>
<td>1. Instructions for Continued Amoerorhness Document No. GEN-040311-0119(Revision IR dated May 5, 2004 or later EASA approved revisions). 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>05/09/2007</td>
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<td>EASA.IM.A.S.00763</td>
<td>NORTHWEST COMPOSITES</td>
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<td>FAA STC ST01565SE</td>
<td>BOEING 737-400, 737-300</td>
<td>Stowage bin Extension.</td>
<td>Northwest Composite Master Drawing List Ref.: A07800 Revision C dated 8th June 2005 or later FAA approved revision FAASTC No. ST01565SE dated 9th June 2005.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>17/06/2005</td>
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<td>EASA.IM.A.S.0090</td>
<td>ROCKWELL COLLINS, Inc.</td>
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<td>FAA STC SA859NM</td>
<td>A320</td>
<td>Installation of Video Entertainment System in accordance with FAA approved Hughes-Avicom Index list 60728.</td>
<td>Revision “A”, dated September 24, 1992, or later FAA approved revisions.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>09/06/2004</td>
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<td>EASA.IM.A.S.0094</td>
<td>STAR AVIATION, Inc.</td>
<td>REV 1</td>
<td>FAA STC ST0594AT</td>
<td>500001, 550051, 5550 AND 560</td>
<td>Installation of RVSM Air Data System in accordance with Master Data Listref-34- Y29-041 Rev. 5 dated 06/12/04 or later FAA approved revision.</td>
<td>Flight Manual Supplement R4-157 Rev. Bo later EASA-approved revision. R34-201 Rev. B or later EASA-approved revision.</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, NAS &amp; associated data shall remain valid until surrendered. 2. If transfer of this Certificate is requested under Part 21A,116, the STC will be terminated. 3. Where aircraft have an Emergency Bus Bar...</td>
<td>08/06/2006</td>
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<td>EASA.IM.A.S.01007</td>
<td>GOODRICH CORPORATION</td>
<td></td>
<td>FAA ST001748B</td>
<td>737-400, 737-500, 737-700, 737-200, 737-200C, 737-300</td>
<td>EASA Validation of FAA STC ST001748B for “Installation of Goodrich Corporation Float Switch Transient Suppression Devices” on Boeing 737 Classic Aircraft.</td>
<td>Master Drawing list for the Float Switch Transient Suppression Devices applicable to Boeing 737-200 through -900 Series Airplanes. Revision F, dated 17th August 2004 or later approved revision FAASTC Approval No. 2034-11309 for AMOC to FAA AD 2004-10-04 for 737 Classic Types.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>17/05/2005</td>
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<td>EASA.IM.A.S.01009</td>
<td>HAWKER BEECHCRAFT CORPORATION (HBC)</td>
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<td>FAA STC SA856GW</td>
<td>HAWKER 1000</td>
<td>Installation of Air Distribution System</td>
<td>-FAA STC SA856GW- Arkansas Aerospace Inc. Drawing List 1000-015-3 Rev B dated June 21, 1994 or later approved revision FAASTC Approval No. 1000-0014 dated June 26 1992 or later approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.IM.A.S.01012</td>
<td>DECA AVIATION ENGINEERING Ltd.</td>
<td>A330-322</td>
<td>Installation of a 10 Flight Attendant Seat. Validation of Transport Canada approved limited STC No D-LASA02-095D issue 3. The STC is defined by DECA modification summary MS80383 issue 2. (or later Transport Canada approved revision)</td>
<td>1. The approval of this modification is limited to the Airbus A330-322MSN 087 and 171 only. The limitations and conditions defined in the EASA STC are applicable. A copy of this STC must be enclosed in the files of each modified aircraft. Any alteration of the STC definition must be made with the...</td>
<td>29/03/2005</td>
<td>Active</td>
<td>10422559</td>
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<td>EASA.IM.A.S.01023</td>
<td>GARMIN INTERNATIONAL, Inc.</td>
<td>SA12546W</td>
<td>DIAMOND DA-40</td>
<td>Installation of Garmin G1000 in the DA40 and DA40F airplane</td>
<td>1. Hartzell Propeller STC EASA.IM.A.0154 must be installed. Diamond Aircraft Industries Modifications listed in Garmin Drawing 905-0034-00 Rev 10 or later approved revision must be installed</td>
<td>11/09/2006</td>
<td>Active</td>
<td>10015372</td>
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<td>EASA.IM.A.S.01026</td>
<td>HAWKER BEECHCRAFT CORPORATION (HBC)</td>
<td>SA8346SW</td>
<td>HAWKER 1000</td>
<td>Installation of Honeywell SRZ-8000 Digital Integrated Flight Control System with the speed awareness cue in accordance with Arkansas Aerospace Inc. Drawing List 1000-008</td>
<td>This STC does not give PRNAV approval. RVSM approval CAT II approved</td>
<td>01/02/2007</td>
<td>Active</td>
<td>10015377</td>
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<tr>
<td>EASA.IM.A.S.01042</td>
<td>JET ENGINEERING LLC</td>
<td>FAA STC ST01542LA</td>
<td>DC-9-81/82/83-87, MD-88</td>
<td>JET Engineering Master Drawing List (MDL) No 290-1000, Supplements/Maintenance Manual for the Jet Engineering Jet Nozzles installed on C-9-80 series aircraft. JET-TR-401 Supplemental Structural Repair Manual to the McDonnell Douglas DC-9-80 Series aircraft Structural Repair Manual</td>
<td>1. The installation should not be incorporated in any airplane unless it is determined that the Interrelationship between this installation and/or previously approved configuration/ configuration does not introduce any adverse effect upon the airworthiness of the airplane. The approval of this modification applies to...</td>
<td>06/09/2005</td>
<td>Active</td>
<td>10422560</td>
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<td>STC Holder</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
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<td>System Record</td>
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<tr>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>FAA STC ST015870A</td>
<td>Installation of ACSS TSCAS System according to FAA STC ST015870A dated 20th January 2004.</td>
<td>- ACSS Index of Master Data Lists, Document No. 8900034, Revision &quot;A&quot; dated 3rd September 2004, or later FAA approved revision: ACSS Aircraft Flight Manual Supplement, Document No. 8002391-001, Revision &quot;A&quot; date...</td>
<td>1. According to limitations section in FAA STC ST015870A dated 20th January 2004.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>17/02/2005</td>
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<td>1015377</td>
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<td>ROEAMOUNT AEROSPACE, Inc.</td>
<td>FAA STC ST021750C</td>
<td>Installation of Rosemount Aerospace 8400M Series Cockpit Door Surveillance System FAA STC ST021750C</td>
<td>- CAA-NL report I/16/LTB/ TCP-05-533157</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/10/2005</td>
<td>Active</td>
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<td>ROEAMOUNT AEROSPACE, Inc.</td>
<td>FAA STC ST022030CH</td>
<td>Installation of Rosemount Aerospace 8400M Series Cockpit Door Surveillance System FAA STC ST022030CH</td>
<td>- CAA-NL report I/16/LTB/ TCP-05-531095</td>
<td>1. FAA approved Airplane Flight Manual Supplement Rosemount Aerospace CDSS AFM Supplement to McDonnell Douglas MD-11, Document D0431158, Revision A, or later FAA approved revision is required on board of modified aircraft.2. This STC is approved only for the product configuration as defined in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>17/02/2006</td>
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<td>1002971</td>
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<td>ROEAMOUNT AEROSPACE, Inc.</td>
<td>FAA STC ST020050CH</td>
<td>Installation of Rosemount Aerospace 8400M Series Cockpit Door Surveillance System FAA STC ST020050CH</td>
<td>- Garmin Aviation Services Drawing List 10018110, Revision A dated October 5 2004 or later approved revision: AFM Supplement 1002071, Revision A dated October 7 2004 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>02/04/2009</td>
<td>Active</td>
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<td>ROEAMOUNT AEROSPACE, Inc.</td>
<td>FAA STC ST025000CH</td>
<td>Installation of Rosemount Aerospace 8400M Series Cockpit Door Surveillance System FAA STC ST025000CH</td>
<td>- Garmin Aviation Services Drawing List 10018110, Revision A dated October 5 2004 or later approved revision: AFM Supplement 1002071, Revision A dated October 7 2004 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>18/07/2005</td>
<td>Active</td>
<td>1002971</td>
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<td>HONEYWELL INTERNATIONAL INC.</td>
<td>CAA-NL report I/16/LTB/ TCP-05-533157</td>
<td>Installation of Honeywell H1 Digital Electronic Engine Control (DEEC)</td>
<td>- Revision A dated October 5 2004 or later approved revision: AFM Supplement 1002071, Revision A dated October 7 2004 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>02/04/2009</td>
<td>Active</td>
<td>1015377</td>
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<tr>
<td>B767-300</td>
<td>Installation of passenger seating arrangement</td>
<td>Tr发展规划 in Master Drawing List 04-05 Revision C, dated April 27, 1999 or later FAA approved revisions.</td>
<td>- FAA approved Airplane Flight Manual Supplement document number 0001, revision Original dated 2/2/02 or later FAA approved revision 2/2/02 or later FAA approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>10/03/2005</td>
<td>Active</td>
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<td>WIKKINSON AIRCRAFT SERVICES, Inc.</td>
<td>FAA STC SA101075C</td>
<td>Installation of a 2 direct drive vacuum pump using the installation instruction document number 0001, revision Original dated 2/2/02 or later FAA approved revision 2/2/02 or later FAA approved revision</td>
<td>- FAA approved Airplane Flight Manual Supplement document number 0001, revision Original dated 2/2/02 or later FAA approved revision</td>
<td>1. Airplane Flight Manual Supplement document number 0001, revision Original dated 2/2/02 or later FAA approved revision required.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>08/08/2005</td>
<td>Active</td>
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<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SA007630E</td>
<td>Installation of Vortex Generators on the Wing leading edge and vertical fin surfaces Drawing Package MA2122 Rev.IR dated 25 October 1999, or later approved revision AFM Supplement for PA-30 &amp; PA-39 dated 13 July 2005, Installation Manual MA2123 Rev. IR dated 8 December 1999 or later approved revision.</td>
<td>- FAA approved Airplane Flight Manual Supplement document number 0001, revision Original dated 2/2/02 or later FAA approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>08/08/2005</td>
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<td>STC No.</td>
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<td>EASA.IM.A.S.01060</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC ST01430CH Installation of universal Avionca Systems Corporation Electronic/茁空/ivalence Situation Indicators (EHSI)</td>
<td>Master Data List ECS-200398, revision H, dated 04 November 2005 or later/approved revision</td>
<td>1. Aircraft Flight Manual Supplement ECS 200408 revision 8 dated 15 November 2005 or later approved revision(2) Instruction for Continued Airworthiness ECS 200411, revision B dated 1 April 2005, if the holder agrees to permit another party to use this certification after the product, the holder shall...</td>
<td>20/06/2006</td>
<td>Active</td>
<td>1010203</td>
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<td>EASA.IM.A.S.01061</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST01256W-D SEE EASA APPROVED MODEL LIST Modification of a Collins TPR-720, TPR-800 or TPR-901 Mode S transponder system installation with an ATC/TCAS control panel to an Elementary Surveillance compliant Collins TPR-401 Mode S transponder system installation with a flight ID/AITC/ TCAS control panel in accordance with Collins Drawing Index as...</td>
<td>FAA STC ST01256W-D, Approved model list No ST01256-W-D. Collins Drawing Index RCA-0010-001, Revision E or later approved revisions.</td>
<td>1. In lieu of an AFM Supplement, the required CableFlightID/AITC/ TCAS control panel pilot's guide, as applicable per FAA approved Collins Drawing Number RCA-180-101, &quot;Installation Instructions, Mode S Transponder and Flight ID/ ATC/TCAS control panel in Part 25 aircraft&quot;, revision 1 dated July 23, 20...</td>
<td>06/06/2006</td>
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<td>10142939</td>
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<td>EASA.IM.A.S.01062</td>
<td>ROCKWELL COLLINS, Inc./Air Transport Systems</td>
<td>FAA STC ST01713LA BOEING 747-400 Installation of Rockwell Collins enhanced Total Entertainment System in accordance with FAA STC No ST01713LA issued February 12, 2005</td>
<td>-System List 1309056, Configuration -002, Revision N, dated February 12, 2005 or later FAA approved Revision</td>
<td>1. FAA STC ST01713SE must be installed in conjunction with this installation. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the...</td>
<td>01/03/2005</td>
<td>Active</td>
<td>1011328</td>
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<td>EASA.IM.A.S.01064</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACS)</td>
<td>FAA ST02798AT 767-200, 767-300, 767-309 Installation of Provisions for Terrain and Traffic Collision Avoidance System (TTCA)/T2CAS This STC is a validation of FAA STC ST02798AT.</td>
<td>-EMTEQ Engineering Master Drawing List EMTEQ-025-MDL Revision dated January 27, 2005 or later FAA approved revision-EMTEQ-025-MDL-001. FAA Approved Revision Doc No 8002064-001</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/03/2005</td>
<td>Active</td>
<td>10142940</td>
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<td>EASA.IM.A.S.01065</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACS)</td>
<td>FAA ST01627LA 767-200, 767-300 Installation of T2CAS</td>
<td>-FAA STC ST01627LA- Doc. No. 8900037, Rev A dated 11 May 2004 or later FAA revision-FAA/Doc. No. 8002064-001, Rev ** dated 27 April 2004, or later FAA revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>21/06/2005</td>
<td>Active</td>
<td>10142942</td>
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<td>EASA.IM.A.S.01066</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA8896SW-D CESSNA T210R CESSNA T210N CESSNA T210M CESSNA T210L CESSNA 201R CESSNA 210N CESSNA 210M CESSNA 210L Installation of S-Tec System SS/SSX Two Axis Automatic Flight Guidance System, Model ST-666, according to Bulletin No 666. Rev 2 dated 1-25-01 or later approved revisions</td>
<td>-S-Tec Master Drawing List No 9234, Rev 8 dated 1-25-01, or later approved revision-FAA/ CAA approved Supplements to pilot's operating handbooks and/or Approved Airplane Flight Manual, as defined in the Limitations and Conditions section of FAA STC SA8896SW-D, latest approved revision Instructions...</td>
<td>1. As defined in the Limitations and Conditions section of FAA STC SA8896SW-D, latest approved revision(2). Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modifications will introduce no adve...</td>
<td>09/09/2008</td>
<td>Active</td>
<td>10113531</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.01067</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>REV. 1</td>
<td>FAA STD STO01628LA</td>
<td>757-200</td>
<td>Installation of T2CAS</td>
<td>FAA STC ST001628LA-A</td>
<td>ACSS Index of Master Data Lists Document Number 8900038 Revision B dated May 10, 2004 or later FAA approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>21/08/2005</td>
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<td>EASA.IM.A.S.01069</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC STO1220LA-D</td>
<td>GULFSTREAM GV S/N 641</td>
<td>Aircraft Serial No. 641 completion and cabin interior installation.</td>
<td>Installation in accordance with approved Gulfstream Aerospace Corporation Index List GC52 0009277, Revision &quot;D&quot;, dated May 13, 2003 or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement C-7517 AFMS, latest approved revision.</td>
<td>1. GV Serial No. 641 only</td>
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<td>17/07/2007</td>
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<td>10015382</td>
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<tr>
<td>EASA.IM.A.S.0107</td>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>CESSNA 177B</td>
<td>CESSNA 177A</td>
<td>CESSNA 177</td>
<td>Installation of S-TEC Systeem 20/30 Single and Two-axes Automatic Flight Guidance Systems.</td>
<td>Installation of S-TEC Systeem 20/30 Single and Two-axes Automatic Flight Guidance Systems. Model ST-741-20/30. In accordance with Bulletin No 841, dated 11-03-97 and Master Drawing list No 921043, dated 11-03-97 or later FAA Approved revisions as defined in FAA STC SA09295AC-D Flight Manual Supplement...</td>
<td>1. Subject to compliance with the provisions of Part 21 Subpart E, this STC and associated data shall remain valid until surrendered, withdrawn or otherwise terminated. 2. If transfer of this Certificate is requested under Part 21A.116, the STC will be reissued. 3. This STC is approved only for...</td>
<td>10/09/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.01071</td>
<td>VIDEO TECHNOLOGY SERVICES, Inc.</td>
<td>FAA STC STO1008NY</td>
<td>767-200, 767-300</td>
<td>Aircraft Serial No. 641 completion and cabin interior installation.</td>
<td>Installation of an AirCell Iridium ST-3100 Single Channel SATCOM -Duncan Aviation Drawing List DA0211-01R Revision A dated December 27/2002 or later approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>23/03/2005</td>
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<td>10042236</td>
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<td>EASA.IM.A.S.01072</td>
<td>DUNCAN AVIATION, INC.</td>
<td>CESSNA 750</td>
<td>Installation of an AirCell Iridium ST-3100 Single Channel SATCOM</td>
<td>-Duncan Aviation Drawing List DA0211-01R Revision A dated December 27/2002 or later approved revision</td>
<td></td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>18/03/2005</td>
<td>Active</td>
<td>10042490</td>
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<td>EASA.IM.A.S.01073</td>
<td>F. ATLEE DODGE AIRCRAFT SERVICES LLC</td>
<td>FAA STC SA00092SA</td>
<td>PA-22-150, PA-22-160</td>
<td>30.5 Gallon Fuel Tanks manufactured in accordance with Drawing List 30.5GAL TANKS, dated 09/03/1992, and installed in accordance with Installation Instructions No 3258-1, rev 1, dated 10/02/1994 or later approved revisions</td>
<td>-FAA STC No SA00092SA-Installation Instructions No 3258-1-FAA approved Supplemental Flight Manual, dated 16/10/1998, or later FAA Approved Revision</td>
<td>1. FAA approved Supplemental Flight Manual, dated 16 October 1998, or later approved revision is a required part of the installation. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>22/08/2005</td>
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<td>EASA.IM.A.S.01075</td>
<td>AIR RADIO SERVICE GMBH</td>
<td>CESSNA 550B</td>
<td>Upgrade of GPWS to EGPWS</td>
<td></td>
<td></td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>10/02/2005</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>ST00353SE.</td>
<td>GULFSTREAM ELECTRONICS, Inc.</td>
<td>FAA STC ST00353SE</td>
<td>2500-0450-009</td>
<td>Installation of Head-Up Guidance System (HGS) model 2100</td>
<td>FAA STC ST00353SE- FAA STC Master Drawing List No ST00353SE- LiteAir Aviation Products Inc.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>11/02/2005</td>
<td>Active</td>
<td>10045400</td>
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<td>ST00638SE.</td>
<td>LITEAIR 10015387</td>
<td>FAA STC ST00638SE</td>
<td>LAP1001-D01 revision D</td>
<td>Installation of Lightweight Window Plugs.(HGS) model 2100</td>
<td>FAA STC ST00638SE- FAA STC Master Drawing List LAP1001-D01 revision D dated May 01, 2003 or later FAA-approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>22/04/2005</td>
<td>Active</td>
<td>10020462</td>
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<td>ST01091SE.</td>
<td>LITEAIR AVIATION PRODUCTS Inc.</td>
<td>FAA STC ST01091SE</td>
<td>LAP1002-D01 revision IR</td>
<td>Installation of Main Deck Lightweight Window Plugs.</td>
<td>FAA STC ST01091SE- FAA STC Master Drawing List LAP1002-D01 revision IR dated October 22, 1999 or later FAA-approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>26/03/2009</td>
<td>Active</td>
<td>10015380</td>
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<td>ST01205NY.</td>
<td>LITEAIR AVIATION PRODUCTS Inc.</td>
<td>FAA STC ST01205NY</td>
<td>LAP1003-D01 revision A</td>
<td>Installation of upper deck lightweight window plugs</td>
<td>FAA STC ST01205NY- FAA STC Master Drawing List LAP1003-D01 revision A dated June 13, 1998 or later FAA-approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>18/11/2005</td>
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<td>ST01237SE.</td>
<td>LITEAIR AVIATION PRODUCTS Inc.</td>
<td>FAA STC ST01237SE</td>
<td>LAP1004-D01 revision A</td>
<td>Installation of Lightweight Window Plugs.</td>
<td>FAA STC ST01237SE- FAA STC Master Drawing List LAP1004-D01 revision A dated July 9, 2003 or later FAA-approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>22/04/2005</td>
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<td>ST01376LA-D</td>
<td>GULFSTREAM AVIATION CORPORATION</td>
<td>FAA STC ST01376LA-D</td>
<td>LAP1005-D01 revision IR</td>
<td>Installation of Main Deck Lightweight Window Plugs.</td>
<td>FAA STC ST01376LA-D- FAA STC Master Drawing List LAP1005-D01 revision IR dated May 01, 2003 or later FAA-approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>22/04/2005</td>
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<td>FAA STC ST01376LA-D</td>
<td>LAP1006-D01 revision IR</td>
<td>Installation of Main Deck Lightweight Window Plugs.</td>
<td>FAA STC ST01376LA-D- FAA STC Master Drawing List LAP1006-D01 revision IR dated November 02, 2004 or later FAA-approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>22/04/2005</td>
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<td>GULFSTREAM AVIATION CORPORATION</td>
<td>FAA STC ST01376LA-D</td>
<td>LAP1007-D01 revision IR</td>
<td>Installation of Main Deck Lightweight Window Plugs.</td>
<td>FAA STC ST01376LA-D- FAA STC Master Drawing List LAP1007-D01 revision IR dated May 01, 2003 or later FAA-approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>FAA STC ST01376LA-D</td>
<td>LAP1008-D01 revision IR</td>
<td>Installation of Main Deck Lightweight Window Plugs.</td>
<td>FAA STC ST01376LA-D- FAA STC Master Drawing List LAP1008-D01 revision IR dated May 01, 2003 or later FAA-approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>ST015087</td>
<td>Gulfstream Aerospace Corporation</td>
<td>B747-400</td>
<td>Installation of Guscapanas External Camera System</td>
<td>FAA STC ST015087A and associated documentation</td>
<td>Gulfstream Flight Manual Supplement GC51901M007</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>25/01/2005</td>
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<td>ST015088</td>
<td>Gulfstream Aerospace Corporation</td>
<td>B747-400</td>
<td>Installation of Airshow CMS Two System</td>
<td>FAA STC ST015088A and associated documentation</td>
<td></td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>25/01/2005</td>
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<td>ST015089</td>
<td>Learjet Inc.</td>
<td>PA-32-260</td>
<td>Installation of an Executive Interior in accordance with Master Drawing List HC200355000 Revision D or later FAA approved revision</td>
<td>FAA approved AFM</td>
<td></td>
<td>1. The Maintenance Instructions for Continued Airworthiness contained in Document Number HC200355000-IFCA-1. Modification of anarcraft by this STC obligates the aircraft operator to include the maintenance information provided by this document in the operator's Aircraft Maintenance Manual and log.</td>
<td>24/01/2005</td>
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<td>10015302</td>
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<td>ST015090</td>
<td>Delta Engineering Corporation</td>
<td>747-400</td>
<td>Description of Design Change Installation of Interior Equipment</td>
<td>FAA STC ST015090A and associated documentation</td>
<td>Delta Engineering Master Data List 2005 2-27121777</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>01/11/2007</td>
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<td>ST015091</td>
<td>Hartzell Service Center</td>
<td>PA-32-260</td>
<td>Installation of Hartzell HC-C2YK-1F/ F7693F three bladed propeller on the Piper PA-32-260 per FAA STC SA1105CH</td>
<td>FAA approved AFM supplement</td>
<td></td>
<td>1. ESA does not consider this modification to increase thermoil level and it was not considered an acoustical change from the HC-C2YK-1F/ F8477-2 Propeller. 2. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written...</td>
<td>30/05/2005</td>
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<td>ST015092</td>
<td>Northwest Aerospace Technologies LLC</td>
<td>757-200</td>
<td>Installation of ELS/EHS S system Upgrade</td>
<td>FAA approved AFM supplement</td>
<td></td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>15/06/2005</td>
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<td>ST015093</td>
<td>STG Aerospace Inc/AKA SaftGlo Inc.</td>
<td>747-400</td>
<td>Installation of SaftGlo Inc. Photoluminescent door proximity emergency escape path marking system (PFEPEMS) in accordance with SaftGlo MDL Report M39-13 rev. B dated October 7, 2004 or later FAA-approved revision</td>
<td>FAA approved AFM</td>
<td></td>
<td>1. The photoluminescent PFEPEMS is only eligible for installation on aircraft model numbers with interior arrangements approved in the SaftGlo MDL, incorporating SaftGlo drawing SG8474 rev B dated October 7, 2004 or later FAA-approved revision. 2. Electrically-powered exit marker lighting is...</td>
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<td>EASA.IM.A.S.01099</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA00582AC-D</td>
<td>AVIATION AIRCRAFT A-1B AVIATION AIRCRAFT A-1A AVIATION AIRCRAFT A-1</td>
<td>Installation of S-TEC System 30 ALT Altitude Hold System, Model ST-881</td>
<td>- FAA STC SA00582AC-D Bulletin No 681 Revision 1, dated September 14, 2000 and Master Drawing List no 921193, Revision A, dated September 14, 2000 or later FAA approved revision (1428 Volt Systems) FAA/DAS Approved Airplane Flight Manual Supplement, Part Number 852004 dated September 19, 2000. 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>01/02/2005</td>
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<td>EASA.IM.A.S.01101</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>ST02126CH</td>
<td>BOEING 757</td>
<td>Installation of elementary &amp; enhanced Mode S Transponders</td>
<td>Electronic cable Specialists Master Data List ECS-202731 revision C dated March 4 2005 or later approved revision ANY Supplement as listed in Electronic Cable Specialists Master Data List ECS-202731 revision C dated March 4 2005 or later approved revision</td>
<td>26/07/2005</td>
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<td>EASA.IM.A.S.01109</td>
<td>AERO ASSOCIATES, Inc.</td>
<td>FAA STC SA00548AT</td>
<td>172 SERIES</td>
<td>Camera Hole Installation</td>
<td>Aero Associates Inc Drawing ED94003 no revision dated march25, 1994 or later FAA approved revision as defined in FAA STC SA00548AT</td>
<td>30/06/2005</td>
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<td>EASA.IM.A.S.01110</td>
<td>BLACKHAWK MODIFICATIONS Inc</td>
<td>FAA SA103415C</td>
<td>B. C90, E90 A. C90A</td>
<td>Installation of Pratt &amp; Whitney Canada PT6A-135A (w/d-rated to 550 HP) engines and/or Raytheon C90, C90A or E90 airplane with Pratt &amp; Whitney Canada PT6A-135A engine configurations shall be determined by the installer. Compatibility with other aircraft/engine configurations defined in the approved design data referred to in the paragraph &quot;Description&quot;. Any defects found must be repaired by approved repair scheme. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>AFMS No 19005 Rev IR; dated 24 November 2004 or later approved revision/Note: approved revision as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Any defects found must be repaired by approved repair scheme. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
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<td>EASA.IM.A.S.01111</td>
<td>LANDMARK AVIATION</td>
<td>FAA STC ST451CH-D</td>
<td>CESSNA 650</td>
<td>Installation of an AlliedSignal Avionics MK-VI Enhanced Ground Proximity Warning System</td>
<td>- FAA STC ST451CH-D Garrett Aviation Services drawing list DL514 revision B dated August 12 1998 or later approved revision/Note: approved revision as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>24/03/2005</td>
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<td>EASA.IM.A.S.01112</td>
<td>LANDMARK AVIATION</td>
<td>FAA STC ST451CH-D</td>
<td>CESSNA 650</td>
<td>Installation of a Bendix/King CAS 67A TCAS II System with Dual Bendix/King MST-67A Mode S Transponders</td>
<td>- FAA STC ST451CH-D Garrett Aviation Services drawing list DL5142 no revision dated January 19 1999 or later approved revision/Note: approved revision as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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| EASA.IM.A.S.01113 | AVIATION PARTNERS, INC. | FAA ST01411SE | - | HAWKER 800XP BAE.125 SERIES 800A/800B | Installation of Winglets, associated changes in wing and Aileron structures and related systems changes on Raytheon Aircraft Company Model Bae.125 Series 800A, 800B and Hawker 800XP Aircraft, in accordance with FAA STC ST01411SE | Installation of this STC on Model Hawker 800XP is limited to Aircraft equipped with the Honeywell DFZ-820 (SPZ-820 avionics) autopilot, the Collins 85/86 (FCC-86E avionics) autopilot and Collins Pro line 21 (FCC-300 avionics) autopilot, for Category I approaches only/Approach
to approaches only/Approach
Airworthiness Limitations. APH-062.. | 26/10/2005 | Active | 10107402 |
<p>| EASA.IM.A.S.01114 | ROCKWELL COLLS, Inc. | B737-400, B737-700, B737-800 | - | Installation of a Video Entertainment System | Sony Trans Com Inc. Index List 982-018B, Rev New, or later FAA approved revisions | 1. B737-400: For Certification basis refer to FAA TCDS A169/plus JAA TGL No.17, IFE Systems1, B737-700, -800: For Certification basis refer to EASA TCDS JAA/2597/418 plus JAA TGL No.17, IFE Systems. This STC is approved only for the product configuration as defined in the approved design data referred to in th.. | 07/04/2005 | Active | 10042953 |
| EASA.IM.A.S.01112 | AIRWORKS | FAA STC ST01433 | A340-211/3J 1-313/642 | This STC deals with the installation of a Cockpit Door Surveillance System. This STC is the validation of the FAA approved STC No ST01433. All documents related to this STC are listed in the Index List Document No 253005 revision A dated March 28th 2004 or later EASA approved revision. | 1. The approval of this modification is limited to the AirbusA340-211/3J1-313/642.. The limitations and conditions defined in the JAA Datasheet are applicable. A copy of this STC and of this supplemental data sheet must be enclosed in the files of each modified aircraft. Any alteration of the STC data... | 12/06/2004 | Active | 10043109 |
| EASA.IM.A.S.01123 | GENERAL AEROSPACE INC | FAA STC ST02820T | BOEING 767-300 SERIES BOEING 767-200 SERIES | Installation of a honeywell RESCU 406 AP ELT system | General Aerospace Master Drawing List SIG/75-ELT- MDL revision D dated April 19, 2005 or later FAA approved revision | 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer. | 30/05/2005 | Active | 10101403 |
| EASA.IM.A.S.01124 | FLIGHT TEST ASSOCIATES, INC. | FAA STC ST01524A | LEARJET 36A | Installation of Rosemount Pilot Static Probe Part Numbers 856N41 &amp; 856N42 | West Star Aviation Inc. Master Drawing List 36A21001 Revision G dated May 20, 2004 or later approved revision, for the airplanes equipped with J.E.T. Model FC-200 Autopilots APH, Supplement 36A04002 Revision B dated August 10, 2004 or later approved revision. Instructions for continued airworthiness... | 05/07/2005 | Active | 10101410 |</p>
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<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<td>FLIGHT TEST ASSOCIATES, INC</td>
<td>FAA STC</td>
<td>ST01525LA</td>
<td>LEARJET 36A</td>
<td>Installation of Honeywell A2-DS2 air data computer. Ba-250 barometric altitude display. AM-250 barometric altimeter. AL-800 altitude alert controller. and Aerodynamic standby altimeter.</td>
<td>West Star Aviation Inc. Master Data List 38A2/20002 revision dated July 12 2004 or later approved revision. For airplanes equipped with J.E.T. models FC-200 autopilots Flight Test Associates AFM supplement 30A04002 revision B dated August 12 2004 or later approved revision. Instructions for costs.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA/IM.A.S.01126</td>
<td>FLIGHT TEST ASSOCIATES, INC</td>
<td>FAA STC</td>
<td>ST01525LA &amp; ST01525LA</td>
<td>LEARJET 35A, 36A, 36B</td>
<td>Aircraft modifications for establishing the approval basis for Reduced Vertical Separation Minimum.</td>
<td>FAA STC ST0152LA &amp; ST01525LA dated August 12 2004 or later approved revision. Learjet models 35/36/36QA with J.E.T. models FC 200 &amp; FC 350 autopilots AFM Supplement 30A04003 revision B dated August 10 2004 or later approved revision. Instructions for continued airworthiness for airplanes equipped with...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA/IM.A.S.01128</td>
<td>AMSAFE, Inc.</td>
<td>FAA STC</td>
<td>SA13050LA</td>
<td>3. 200H, 200H</td>
<td>Installation of AMSAFE Aviation Inflatable Restraint (AIR) System.</td>
<td>-FAA STC No SA01700LA - AMSAFE Inc. Master Data List E208880Revision NC dated November 10, 2004 or later approved revision. - AFM Supplement E508810 Revision IR dated November 10, 2004 or later approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>22/04/2005</td>
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<td>EASA/IM.A.S.01130</td>
<td>STERLING AVIATION TECHNOLOGIES Inc.</td>
<td>FAA STC</td>
<td>SA11306E</td>
<td>PA-38-112</td>
<td>Installation of Tomahawk Fatigue Life Extension Wing Spar Modification. FAA STC SA11306E. Validation is in accordance with EASA Certification Directorate Exposure Part 2, Subpart 3 Internal Working Procedure/Supplemental Type Certification (STCP) Issue 1 adopted 27 June 2000.</td>
<td>- FAA STC SA11306E - Steriing Aviation Installation Instructions SA38-112-000 Revision Dated February 8 2007. - FAA STC SA38-112-000 Revision Dated September 24 2002 or later approved revision.</td>
<td>Approval restricted to modifications carried out in accordance with new installation instructions. This STC is approved only for the product configuration as defined in the app.</td>
<td>08/03/2007</td>
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<td>EASA/IM.A.S.01134</td>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>REV.</td>
<td>FAA</td>
<td>B.C60, E90</td>
<td>Installation of Pratt &amp; Whitney PT6A-135A engine with 1. Hartball propellers HC- D40-3CD92005S.K with a diameter of 50 inches in accordance with Blackhawk Modification, Inc. Drawing No 19003-01, Rev B, dated 1st April 2004, &quot;Installation Instructions&quot;, or later approved revision. Note: asp...</td>
<td>AFMS No 19003-00, Rev A, dated 7th October 2004 or later approved revision</td>
<td>AFMS No 19003-00, Rev A, dated 7th October 2004 or later approved revision is required for the Raytheon C60, C90A or E90 airplane with Pratt &amp; Whitney Canada PT6A-135A engines. Note that the shaft horsepower is limited to 550 at 1900 RPM. Each engine mount must be removed from the airplane, cleaned and...</td>
<td>20/04/2006</td>
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### List of EASA Supplemental Type Certificates

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<td>10042954</td>
<td>NKN AVIATION SERVICES, Inc.</td>
<td>FAA STC ST09943SC</td>
<td>FAR/CHILD DORMER DO 328-300</td>
<td>FAA STC ST09943SC/12 Seats configuration extended to 14 seats through minor change, part of Aircraft OH-P 0353/Installation of interior cabin, cabinetry, seats, card tables, monitors, entertainment system, system wiring, and lavatory in accordance with AvCraft Completions Master Document List MD...</td>
<td>FCOM Temp Rev No 05-11, effective for S/N 3199, FAA Doc No AFMS-D03-011575, Revision IN dated June 10, 2002 or later FAA approved revision</td>
<td>1. This STC may be installed only in combination with Change Notice CN-F10535 - F10404 -F10125 owned by Aircrft Germany and RUAG minor Changes “In Seat Power Supply and F1 Switch” and “Installation of Double Latches on drawers” 2. The water pumping system for the galley and the lavatory sink are i...</td>
<td>20/06/2006</td>
<td>Active</td>
<td>10011742</td>
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<td>10015418</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC ST06546WI</td>
<td>LEARJET 60</td>
<td>Installation of a Universal Avionics Inc. UNS-18 FMS (Dual)</td>
<td>FAA STC ST00646WI - Duncan Aviation Inc. Document List No1004830, May 27, 1997, or later FAA approved revisions - Duncan Aviation Inc. Document No 4017300, FAA-approved AFMS June 13, 1997, or later FAA approved revisions.</td>
<td>1. For certification basis refer to FAA TCDS A10CE.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” Compatibility with other aircraft engine configurations shall be determined by the installer.</td>
<td>11/04/2005</td>
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<td>10015419</td>
<td>WIPARE Inc.</td>
<td>FAA STC SA00900CH</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of Wirapane Model 2350 Amphibian/Seaplane floats on CessnaT2 Models (validation of FAA STC SA00900CH)</td>
<td>Installation in accordance with Wirapane Inc. MDL WI-MD010 issue date 25July, 2005 or later approved revisions.</td>
<td>1. Installation on Cessna models 172R and 172S require aircraft stretched McCauley Model IA200/WFA propeller and Textron Lycoming ModelIO-360-L2A engine rated at 180hp and 2100 rpm.2. FAA STC SA332GL is required for models 172I,K,L,M,N and P (Note: FAASTC will require FAA approved AFMS, June 13, 1997, or later FAA approved revisions.</td>
<td>04/07/2005</td>
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<td>10015411</td>
<td>AVIATION TECHNICAL SERVICES, Inc.</td>
<td>FAA STC ST00886E-D</td>
<td>757-200</td>
<td>Installation of an Interior Arrangement, Altronics &amp; Ozone Converters.</td>
<td>FAA STC No ST00988SE-D-Goedrich Master Drawing List 2004-005 Revision A, dated May 4 2004 or later approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft engine configurations shall be determined by the installer.</td>
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<td>10015412</td>
<td>S-TEC CORPORATION</td>
<td>FAA SA091D0AC-D</td>
<td>PILATUS PC 6 SERIES</td>
<td>Installation of Miggitt Avionics S-TEC Autopilot 555XK</td>
<td>Master Drawing List ST-558 PIN 28264 Rev C or later approved revisions CWI No 55603 applies for control stick modification/Airplane Flight Manual Supplement S-TEC PIN 8912295 rev 2or later approved revision/Instructions for Continued Airworthiness P N 81581 Rev 2 or later approved revision</td>
<td>1. Limited to PC-6B12-12; PC-6/ B2-H2; PC-6B2-H4 and PC-6/ B2-H4 when modified with underwing fuel tanks.2. If stabilization/automatic pitch trim switch S943 is installed, the optional trim wiring in accordance with detail “A” on S-TEC drawing No 99213 Rev C or later approved revisions must be insta...</td>
<td>10/06/2008</td>
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<td>EASA.IM.A.S.01150</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0175LA-D</td>
<td>GULFSTREAM GH-5P</td>
<td>Gulfstream Aerospace Corporation Modification FAA STC ST 01752 LA.D. Aircraft Completion &amp; Custom Cabin Installation. Gulfstream DAS Report Number GC52000K329 Dated 03/08/05 Flight Manual Supplement ref GC 52000M012 Dated 03/08/05 or later approved revisions</td>
<td>Gulfstream DAS Report Number GC52000K329 dated 03/08/05 Flight Manual Supplement ref GC 52000M012 dated 03/08/05 or later approved revisions</td>
<td>1. Serial 5068 only: This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.IM.A.S.01151</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST01959NY</td>
<td>MD-88 DC-9-87 (MD-87)</td>
<td>Validation of FAA STC No ST01959NY; Installation of Dual Transponders with ACSS 7517800-10005 ATDL Transponders with ELTs and Gobles G7490-58 Control Head.</td>
<td>FAA STC No ST01959NY-Delta Engineering MSL/D 0158-0637-2076 Rev. A dated 30 April 2004.</td>
<td>1. This installation does not include EHSI or ADS-B functionality. The Delta Engineering MML No 0758-05312-2076 dated 19 April 2004 is required.</td>
<td>22/07/2005</td>
<td>Active</td>
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<td>EASA.IM.A.S.01152</td>
<td>S-TEC CORPORATION</td>
<td>CESSNA 180D, 180E, 180F; CESSNA 180, 180A, 180B, 180C</td>
<td>Installation of S-TEC System 40/50 Single &amp; Two Axis automatic Flight guidance systems, Model ST-196-40/50</td>
<td>Bulletin No. 296 Revision 8 dated March 22, 1999. Master Drawing ListNo. 92232 Revision H dated March 22, 1999 or latest EASA approved revisions. (14 Volt System) S-TEC AFM Supplement/PN 8548 applicable to system 40 dated July, 1998 or later FAA approved revision and UK CAA Change Sheet 101 Issue 1.</td>
<td>This STC applies to Cessna Models 180, 180A, 180B, 180C, 180D, 180E, 180F (Floatplane or amphibian) &amp; the above models when equipped with FAA STC SA1435WE (Roberts STOL) or FAA STC SA067CE (Horton STOLs) and / or FAA STC SA1705WE (InteRally tanks) 2. This STC applies to Cessna Models 180, 180A, 1..</td>
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<td>EASA.IM.A.S.01153</td>
<td>DAVIS AERO ENGINEERING, Inc.</td>
<td>FAA STC ST01857LA</td>
<td>LEARJET 60</td>
<td>Installation of Rockwell Collins TCAS II.</td>
<td>FAA STC ST01657LA. Davis Aero Engineering MSL Doc. No AG362-MLD, Rev. B, or later FAA approved revisions. AFMS AG362-600, Rev. IR, or later FAA approved revisions.</td>
<td>1. For certification basis refer to FAA TCDS A10CE plus JAA/TLG No 8 and 13.</td>
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<td>EASA.IM.A.S.01154</td>
<td>TEXAS SKYWAYS, INC.</td>
<td>FAA STC SA09133SC</td>
<td>CESSNA F162-2, CESSNA F182-P</td>
<td>Installation of Teledyne Continental Motors IO-550-B1 enginew Fireball or McCauley Propeller in acc. with Texas Skyways Spec. TS0028, Rev. 1, or later approved Revision and TS0034</td>
<td>FAA STC SA09133SC (Texas Skyways AFM-S FS550182, Rev. 2, or later approved Revision)Texas Skyways AirCraftainion 0637, dated 27 March 1995, or later approved Revision</td>
<td>The approval is only valid in combination with the installation of STC EASA.IM.A.S.01374 (Gomolzig GmbH, Germany)</td>
<td>07/07/2006</td>
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<td>EASA.IM.A.S.01155</td>
<td>TEXAS SKYWAYS, INC.</td>
<td>FAA STC SA10070SC</td>
<td>CESSNA U206G, CESSNA U206F, CESSNA U206D, CESSNA U206C, CESSNA U206B, CESSNA U206A, CESSNA U206</td>
<td>Installation of Hartzell 3-blade Propeller in acc. with Texas Skyways Doc. 5016-0000, Rev. E, or later approved Revision</td>
<td>FAA STC SA10070SC (Texas Skyways AFM-S 5016-180, Rev. IR, or later approved Revision for Cassina 180 series)</td>
<td>The approval is only valid with latest EASA approved Noise Certificate</td>
<td>07/07/2006</td>
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### List of EASA Supplemental Type Certificates

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<th>Limitations/Conditions</th>
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<td>EASA.IM.A.S.01155</td>
<td>TEXAS SKYWAYS, INC.</td>
<td>FAA STC SA10070SCTexas Skyways AFM-S 5016-180, Rev. IR, or later approved Revision for Cessna 180 series</td>
<td>CESSNA 180 series</td>
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<td>AVIATION FABRICATORS, Inc.</td>
<td>FAA ST00716WI</td>
<td>2. 501, 551 1. 500, 550, 555, 560</td>
<td>Installation of Shoulter Harness Kit</td>
<td>FAA STC ST00716WI</td>
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<td>EASA.IM.A.S.01165</td>
<td>STG AEROSPACE Inc.AKA SAFTGLO, Inc.</td>
<td>FAA STC ST02697AT</td>
<td>BOEING 717-200</td>
<td>Installation of SafTGlo Inc. Photoluminescent floor proximity emergency escape path marking system (FPEEPMS)</td>
<td>Cabin and exit lighting levels must be measured for acceptability on each installation in accordance with STG Aerospace Report No. P51-12. Revision A dated November 13, 2002 and P51-10. Revision I.R., dated August 15, 2002 or later approved revisions</td>
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<td>STG Holder</td>
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<td>STG AEROSPACE Inc AKA SAFTGLO Inc.</td>
<td>DOUGLAS DC-9-33, DC-9-34, DC-9-31 (NOT CERTIFIED)</td>
<td>Installation of a photoluminescent floor proximity emergency escape path marking system.</td>
<td>- Flight Manual Supplement AFMS DAS Report No. H75-11 Rev. 1(Rated 7 July 1999 - Dispatch Operating Procedures) DAS Report No. H75-13 Rev. C dated 7 July 1999 or later approved revisions.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>13/02/2009</td>
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<td>SAF-T-GLO Inc</td>
<td>DC-9-51</td>
<td>Installation of a photoluminescent floor proximity emergency escape path marking system.</td>
<td>Flight Manual Supplement AFMS DAS Report No. H65-11R dated 13 September 1999 Dispatch Operating Procedures DAS Report No. H65-13R dated 13 September 1999 or later approved revisions.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>STG AEROSPACE Inc AKA SAFTGLO Inc.</td>
<td>FAA STC ST03041AT</td>
<td>Installation of a photoluminescent floor proximity emergency escape path marking system.</td>
<td>-STG Aerospace MDL Report Number M41-2 Revision D dated November 3, 2004 or later approved revision-AD Design Document No. M41-11, Revision A dated June 24, 2004 or later approved revisions.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>SAT-T-GLO Inc</td>
<td>FAA STC ST02651AT</td>
<td>Installation of a photoluminescent floor proximity emergency escape path marking system.</td>
<td>FAA AFMS DAS Report No. H49-11R dated 7 July 1999 - Dispatch Operating Procedures DAS Report No. H49-13 Rev. C dated 7 July 1999 or later approved revisions.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>STG AEROSPACE Inc AKA SAFTGLO Inc.</td>
<td>FAA STC ST01932AT</td>
<td>Installation of a photoluminescent floor proximity emergency escape path marking system.</td>
<td>FAA AFMS DAS Report No. H65-11R dated 13 September 1999 Dispatch Operating Procedures DAS Report No. H65-13R dated 13 September 1999 or later approved revisions.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>STG AEROSPACE Inc., AKA SAFTOLO, Inc.</td>
<td>FAA STC ST015439</td>
<td>BOMBERDIER DHC-7-102</td>
<td>Installation of SaftGlo Inc. Photoluminescent floor proximity emergency escape path marking system (FPEPMS)</td>
<td>-DAS MDL Report Number G34-2 Revision A dated April 18, 1999 or later approved revisions -AFM Supplement G34-11 dated April 28, 1999 or later approved revisions -Cabin and exit lighting levels must be measured for acceptability on each installation in accordance with DAS Report No. G34-12, Revision A dated April 18, 1999 and G34-10, Revision A, dated April 17, 1999 or later approved revisions.</td>
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<td>FAA STC ST015440</td>
<td>REV 2</td>
<td>Supplemental Operating Configurations MD-50-Addition of engine and part intermix.</td>
<td>MDL AA 1781 (Rev. H) AFM Supplement</td>
<td>26/10/2007</td>
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<td>ST03041AT</td>
<td>JET AVIATION AG, BASEL</td>
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<td>AIRBUS A319-115</td>
<td>Cabin interior outfitting</td>
<td>FAA STC No BC02507-201-01</td>
<td>Flight Manual Supplement BC02507-201-01</td>
<td>Approval valid for S/N 2507 only. The aircraft is certified for private use only. It is prohibited to operate the aircraft for hire or offered for common carriage. The aircraft is NON SMOKING aircraft. For other limitations, see flight manual supplement BC02507-201-01.</td>
<td>20/12/2005</td>
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<td>ST03042AT</td>
<td>JET AVIATION AG, BASEL</td>
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<td>A320-232</td>
<td>Cabin Modification (Final Completion of VIP Configuration)</td>
<td>FAA STC No BC02506-201-01</td>
<td>Flight Manual Supplement Doc. No BC02506-201-01</td>
<td>1) Modification limited to Aircraft S/N 25062. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>21/12/2005</td>
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<td>JET AVIATION AG, BASEL</td>
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<td>LEARJET 60 LEARJET 55C LEARJET 55 LEARJET 56A LEARJET 36 LEARJET 35A LEARJET 35 LEARJET 31A LEARJET 31</td>
<td>Activation of the Runway Awareness and Advisory System (RAAS)</td>
<td>Master Data List Doc No BLJ1334461-001FOCA approved AFM Supplement Document No BLJ1334461-201-01 Rev IR</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>30/05/2005</td>
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<td>ST03044AT</td>
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<td>LEARJET 55C LEARJET 55 LEARJET 56A LEARJET 57 LEARJET 36 LEARJET 35A LEARJET 35 LEARJET 31A LEARJET 31</td>
<td>Installation of Transponder Flight ID capable Mode S Transponders and Flight ID capable Transponder Control Panel</td>
<td>FAA STC No ST02976AT dated 10th November 2004/ASG Master Drawing List, Ref 24640 Rev A dated 9th February 2004 and/or FDR Master Data List No MDL96060 Initial Revision, dated January 10, 1997 or later FAA approved revision.</td>
<td>1. According to FAA STC No ST02976AT. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>25/10/2005</td>
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<td>ST03045AT</td>
<td>AVIONICS SUPPORT GROUP, Inc</td>
<td>FAA STC 737-300, 737-400, 737-500</td>
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<td>Installation of ACCS Elementary and Enhanced Surveillance capable Mode S Transponders and Flight ID capable Transponder Control Panel</td>
<td>FAA STC No ST02976AT dated 10th November 2004/ASG Master Drawing List, Ref 24640 Rev A dated 9th February 2004 and/or FDR Master Data List No MDL96060 Initial Revision, dated January 10, 1997 or later FAA approved revision.</td>
<td>1. FAA approved Airplane Flight Manual Supplement ADN958060 dated 18th APR 1997 or later approved revision. Is a requirement of this STC. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>31/08/2005</td>
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<td>ST03046AT</td>
<td>RAYTHEON AIRCRAFT SERVICES, INC</td>
<td>FAA STC 737-300, 737-400, 737-500</td>
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<td>Installation of the Runway Awareness and Advisory System (RAAS)</td>
<td>FAA STC No ST02976AT dated 10th November 2004/ASG Master Drawing List, Ref 24640 Rev A dated 9th February 2004 and/or FDR Master Data List No MDL96060 Initial Revision, dated January 10, 1997 or later FAA approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>28/06/2005</td>
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<td>ST03047AT</td>
<td>GULFSTREAM AEROSPACE Corp</td>
<td>FAA STC</td>
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<td>Installation of Transponder Flight ID</td>
<td>Gulfstream Aerospace Corporation Index List CEB11268000, Revision NC Dated 24 June 2004, or later FAA approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>ST03048AT</td>
<td>GENERAL AEROSPACE INC REV. 1</td>
<td>FAA STC</td>
<td></td>
<td>Installation of Honeywell Aircraft Communication Addressing &amp; Reporting System (ACARS).</td>
<td>General Aerospace Master Drawing List No. GA67-ACARS-MDL Rev. B</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>30/05/2005</td>
<td>Active</td>
<td>10010451</td>
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<td>EASA STC No.</td>
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<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM/AS.01196</td>
<td>Gulfstream Aerospace Corporation</td>
<td>ST0153LAD</td>
<td></td>
<td>Gulfstream GV-SP</td>
<td>Aircraft completion and cabin interior installation</td>
<td>Gulfstream DAS report No. GC52000M4 dated 13 SEP 2005 or later approved revisions.</td>
<td>This STC is limited to Gulfstream GV-SP Serial Number 5070 only.</td>
<td>21/02/2005</td>
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<td>EASA.IM/AS.01198</td>
<td>Lifeport, LLC</td>
<td>FAA STC SA010358SE</td>
<td></td>
<td>Cessna 172</td>
<td>Installation and inspection of emergency medical stretcher and base; oxygen system in accordance with FAA STC SA010358SE or later approved revision.</td>
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<td>EASA.IM/AS.01199</td>
<td>Lifeport, LLC</td>
<td>FAA STC ST00990SE</td>
<td>LEARJET 45</td>
<td>Installation of emergency medical stretchers and bases, oxygen and electrical systems and associated Lifeport Inc. Report No. EGR-644-12 Revision IR dated November 6 2000 &amp; Master Drawing List EGR-644-00 revision R dated 25 April 2001 or later approved revision</td>
<td>Lifeport Inc. Report No. EGR-644-12 Revision IR dated November 6 2000 &amp; Master Drawing List EGR-644-00 revision U dated 1 January 2005 or later EASA approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>21/06/2005</td>
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<td>10/1544</td>
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<td>EASA.IM/AS.01201</td>
<td>S-TEC Corporation</td>
<td>FAA STC SA02693AC-D</td>
<td>Navion H</td>
<td>Installation of S-TEC System 20/30 Single and Two Axis Automatic Flight Guidance System Model ST-724/20/30 in accordance with FAA STC SA02693AC-D</td>
<td>-FAA/DAS approved Supplemental Flight Manual P/N 891353, Rev 1 for Navion models A, B, D, E, F, G and H when modified by aho Alto Airport Inc. or Symor's Engineering Horizontal Stabilizer Angle of Incidence Change with or without tip tanks and the above models when modified by 285 H.P. engine conversion. 1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.IM/AS.01202</td>
<td>Gulfstream Aerospace Corp.</td>
<td>FAA STC ST102965C-D</td>
<td>Gulfstream 200</td>
<td>Installation of ICG-200 SATCOM System</td>
<td>-Index List GA211039000, Revision B dated 15 February 2005 or later FAA approved revision FAA approved Airplane Flight Manual Supplement doc. Number GA211039000 Rev none dated 19 February 2005 or later FAA approved revision</td>
<td>1. Instructions for continued Airworthiness GA214034000 Revision N/C dated 31 January 2005 or later FAA approved revision is required. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other...</td>
<td>28/06/2005</td>
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<td>10/1546</td>
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<td>EASA.IM/AS.01203</td>
<td>Gulfstream Aerospace Corp.</td>
<td>FAA STC ST102975C-D</td>
<td>Gulfstream 200</td>
<td>Installation of Executive Cabin Interior</td>
<td>-Gulfstream Aerospace Index List GA232068103, Rev A dated 14 February 2005 or later FAA approved revision FAA approved Flight Manual Supplement GA242684103, Rev none dated 14 February 2005 or later FAA approved revision</td>
<td>1. This STC is limited to the installation made in Gulfstream Aerospace L(F) (Israel Aircraft Industries, LTD) model Gulfstream 200 serial number 101-2. This STC does not permit manufacturing of part for multiple installation3. This STC is approved only for the product configuration as defined...</td>
<td>28/06/2005</td>
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<td>10/1547</td>
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<td>EASA.IM/AS.01204</td>
<td>Gulfstream Aerospace Corp.</td>
<td>FAA STC ST102965C-D</td>
<td>Robin DR 400 RP, DR 120D, DR 140B, DR 180R, DR 200R</td>
<td>Installation of WX-1000E Stormscope System</td>
<td>-Gulfstream Index List GA212125800 Revision D dated 17 February 2005 or later FAA approved revision FAA approved Flight Manual Supplement GA212125800 Rev None dated 17 February 2005 or later FAA approved revision</td>
<td>1. This installation is eligible for airplanes in S/N 052 and subsequent with Collins Pro Line 4, version 6.1 avionics system2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft...</td>
<td>28/06/2005</td>
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<td>10/1548</td>
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<td>EASA.IM/AS.01205</td>
<td>Gulfstream Aerospace Corp.</td>
<td>FAA STC ST102965C-D</td>
<td>Gulfstream 200</td>
<td>Installation of Honeywell LaserIF V IRS</td>
<td>-Index List GA211138000, Revision B dated 18 August 2004 or later FAA approved revision FAA approved Airplane Flight Manual Supplement doc. Number GA211138000 Rev None dated 18 August 2004 or later FAA approved revision</td>
<td>1. Compliance with FAA Special Condition 25.263-6C and AC/GC/ EASA CRI F-08 for HRFV has been demonstrated. 2. The aircraft must be equipped with Rockwell Collins FMS-8100 Flight Management Systems installed in accordance with EASA MOD Nos 7219and 100233. This STC is approved only for the product...</td>
<td>28/06/2005</td>
<td>Active</td>
<td>10/1549</td>
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<td>EASA STC no.</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.01206</td>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA STC ST010205C-D</td>
<td>GULFSTREAM 200</td>
<td>Installation of DeCrane/EDRGA Cabin Double Seats</td>
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<td>EASA.IM.A.S.01207</td>
<td>MCFARLANE AVIATION, Inc.</td>
<td>FAA STC SA01074WI</td>
<td>CESSNA 207, 210, CESSNA 180, 182, 185, 188, 206, CESSNA 150, 152, 170, 172, 175</td>
<td>Installation of McFarlane Aviation FlightSafety Assembly Service Kit MCKS100 under FAA STC SA01074WI</td>
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<td>EASA.IM.A.S.01209</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>FAA STC ST01256AT</td>
<td>BOEING 767-300 SERIES</td>
<td>Installation of Flight Data Acquisition &amp; Management System (FDAMS) with Mark II Aircraft Communication &amp; Reporting System (ACARS)</td>
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<td>EASA.IM.A.S.01210</td>
<td>JET AVIATION AG, BASEL</td>
<td>-</td>
<td>BOMBARDIER CL-600-1A11 (CL-600)</td>
<td>Flight Management System (FMS) Upgrade &amp; Terrain Awareness Warning System (TAWS) Installation</td>
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<td>EASA.IM.A.S.01213</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>-</td>
<td>BEECHCRAFT B300, B300C</td>
<td>Installation of dual Collins TDR-94/94D Mode S Transponder installation of single Collins TDR-94/94D Mode S Transponder and single TDR-90 Mode C Transponder on Beechcraft B300 &amp; B300C</td>
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<td>EASA.IM.A.S.01214</td>
<td>ROCKWELL COLLINS, Inc.d/b/a AIR TRANSPORT SYSTEMS</td>
<td>SA1930CE-D</td>
<td>BEECH B300C BEECH B300</td>
<td>Installation of Collins TCAS-94 TCAS II System.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.IMA.S.01220</td>
<td>BOMBARDIER AEROSPACE</td>
<td>10015468</td>
<td>TCCA C-LSA05-103D</td>
<td>BD-700-1A10</td>
<td>Cabin Interior Completion.</td>
<td>Doc. Limited Supplemental Type Certificate by TCCA No. C-LSA05-103/D. FAA STC SA01513SE dated 9 June 1998, revised on 20 January 2004. 328694 - Installation and Calibration Procedures</td>
<td>1. This STC is limited to Serial Number 914-2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>14/06/2005</td>
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<td>EASA.IMA.S.01221</td>
<td>F. ATLEE DODGE AIRCRAFT SERVICES LLC</td>
<td>10015469</td>
<td>FAA STC SA02115AK</td>
<td>PIPER PA-18S-150 PIPER PA-18A-150 PIPER PA-14 PIPER PA-12</td>
<td>Fabrication and installation of F. Atlee Dodge 30.5 gallon wing fuel tanks sight gages</td>
<td>F. Atlee Dodge Master Data List No. 328694.05 Revision 1, dated 16 January 2004, or later approved revisions. FAA STC SA02115AK dated 9 June 1998.</td>
<td>1. Any Piper aircraft incorporating this change must be equipped with F. Atlee Dodge 30.5 gallon wing fuel tanks in accordance with FAA STC SA0449A1, SA00002SE, or SA0781NM. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”.</td>
<td>06/06/2005</td>
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<td>EASA.IMA.S.01222</td>
<td>CASCADE AVIATION SERVICES INC</td>
<td>10015499</td>
<td>SA01513SE</td>
<td>BAE JETSTREAM 320I SERIES</td>
<td>Installation of Sandel Terrain Awareness Warning System (TAWS)</td>
<td>FAA STC SA01513SEMDL: 04CAS032B-ML, Rev. 04CAS032-DB, Rev C</td>
<td>Limitations as per Flight Manual</td>
<td>05/02/2007</td>
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<td>EASA.IMA.S.01223</td>
<td>GENERAL AEROSPACE INC</td>
<td>10015500</td>
<td>FAA STC ST03002AT</td>
<td>BOEING 767-300/767-200</td>
<td>Installation of a Honeywell Enhanced Ground Proximity Warning System.</td>
<td>General Aerospace Master Drawing List No. GA672: EGUPS-MDL. Revision B, dated May 24, 2005 or later EASA approved Revisions. General Aerospace AFM Supplement GA37: EGUPS-AFM501 dated December 10, 2004 or later EASA approved revisions.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>14/06/2005</td>
<td>Active</td>
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<td>EASA.IMA.S.01224</td>
<td>BEECHCRAFT CORPORATION</td>
<td>21/05/2008</td>
<td>FAA STC ST10385SC-D</td>
<td>HAWKER 850XP, HAWKER 900XP</td>
<td>Installation of a Rockwell Collins CMS-1 Cabin Management System.</td>
<td>1. In accordance with Master Drawing List 800-512 Revisions AB, or later approved revisions. 2. Flight Manual Supplement 800-319 Revisions R9 or later approved revisions. 3. Maintenance Manual Addenda 33-10-10, 33-20-15, 33-30-02.</td>
<td>1. This modification requires installation of STC ST10385SC-D. Installation of an Emergency Light System.</td>
<td>21/05/2008</td>
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<td>EASA.IMA.S.01225</td>
<td>BEECHCRAFT CORPORATION</td>
<td>21/05/2008</td>
<td>FAA STC ST10386SC-D</td>
<td>HAWKER 850XP, HAWKER 900XP</td>
<td>Installation of an Emergency Light System.</td>
<td>1. In accordance with Master Drawing List 800-513 Revisions L, or later approved revisions. 2. Flight Manual Supplement 800-317 Revisions RS or later approved revisions. 3. Maintenance Manual Addenda 33-50-05.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>21/05/2008</td>
<td>Active</td>
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<td>EASA STC no.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.MAS.01226</td>
<td>TRANSAIRCO S.A.</td>
<td>-</td>
<td></td>
<td>MYSTERE FALCON 50</td>
<td>Replacement of existing HONEYWELL MK II GPWS and COLLINS MFD-85 with a UNIVERSAL Terrain Awareness and Warning System (TAWS) and a UNIVERSAL MFD-640</td>
<td></td>
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<td>1/09/2005</td>
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<td>EASA.MAS.01229</td>
<td>SAFRAN CABIN BELLINGHAM, INC.</td>
<td>FAA STC</td>
<td>ST015326E</td>
<td>B747-400</td>
<td>Cabin Interior Reconfiguration</td>
<td>Health Tecno Inc. documents for Project PB1 &amp; associated Master Drawing List HPD-DL-93 Rev A dated April 15, 2005 or later FAA approved revision</td>
<td></td>
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<td>30/05/2005</td>
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<td>EASA.MAS.01231</td>
<td>BOMBARDIER INC.</td>
<td>C.LSA05-024/ D</td>
<td>C.BOMBIARDER BD-700-1A11, BOMBARDIER BD-790-1A10</td>
<td>Installation of Thales Integrated Electronic Standby Instruments for BD700-1A11 Global Express and BD710-1A11 Global 5000/Validation of imported TCC STC: SA0A-118</td>
<td></td>
<td>AFM Supplement G- FM34004001 Rev B or later TCCA approved rev.</td>
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<td>08/08/2005</td>
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<td>Rev. level</td>
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<td>Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.01238</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>FAA STC</td>
<td>103956C</td>
<td>MD-11</td>
<td>EASA Validation of FAA STC ST103956C on &quot;Installation of two non-tilt, lower-lifting, &quot;ITTC&quot; manual control yoke and associated Aircraft Flight Manual Supplement&quot;</td>
<td>FAA STC ST103956C dated 1 April 2005, FAA STC ST029236C dated 1 April 2005, FAA STC ST034237 dated 1 April 2005</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>31/05/2005</td>
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<td>EASA.IM.A.S.01240</td>
<td>L3 AVIATION PRODUCTS, INC</td>
<td>FAA STC</td>
<td>01837AT</td>
<td>757-200</td>
<td>Additional Parameters for Flight Recording Systems: Air. Flight System Report 98-10-24 Revision C dated December 2004 or later approved revision</td>
<td>Flight Systems Report 98-10-24 Revision C dated December 2004 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td></td>
<td>29/11/2005</td>
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<td>EASA.IM.A.S.01241</td>
<td>AEROSPACE GMBH</td>
<td>FAA STC</td>
<td>09468AC</td>
<td>DORNIER 325-100</td>
<td>Modification of the cabin floor to be the same level through out the cabin, according to FAA STC No ST09468AC, reissued 18 April 2005</td>
<td>-Fairchild Aircraft Report R1530, Revision D dated 20 April 1999-FAA STC ST09468AC, reissued 18 April 2005</td>
<td>1. According to the application as forwarded by FAA letter dated 27 April 2005, the approval is limited to the aircraft S/N 30952. Furthermore the limitations as referenced in FAA STC ST09468AC, reissued 18 April 2005, apply. This STC is approved only for the product configuration as defined in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td></td>
<td>15/06/2005</td>
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<td>EASA.IM.A.S.01242</td>
<td>KNH AVIATION SERVICES, Inc.</td>
<td>FAA STC</td>
<td>09468AC</td>
<td>DORNIER D328-100</td>
<td>Installation of a corporate/executive interior</td>
<td>-Fairchild Aircraft Engineering Change Package (ECP) 3000, Revision A dated 2005/06/30-FAA STC ST09468AC, reissued 18 April 2005</td>
<td>1. According to the application as forwarded by FAA letter dated 27 April 2005 and FAA STC ST09468AC, the approval is limited to the aircraft S/N 30952. Furthermore the installation has to be performed in combination with the installation of FAA STC ST09468AC and the limitations as referenced in FAA STC ST09468AC.</td>
<td></td>
<td>15/06/2005</td>
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<td>EASA.IM.A.S.01243</td>
<td>DEVOIE AVIATION CORPORATION OFAMERICA</td>
<td>FAA STC</td>
<td>09191AC</td>
<td>BOEING 737-500</td>
<td>Installation of DaVine V-200 Forehead Position Light Assembly according to FAA STC ST09191AC</td>
<td>FAA STC No ST09191AC, Date reissued 5 May 2005-FAA STC ST09191AC, Date reissued 6 May 2005</td>
<td>1. According to Limitations Section of FAA STC ST09191AC, the STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td></td>
<td>25/10/2005</td>
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<td>EASA.IM.A.S.01244</td>
<td>FLIGHT ENVIRONMENTS INC</td>
<td>FAA STC</td>
<td>01318LA</td>
<td>BEECH 390</td>
<td>Installation of thermal/acoustic insulation kit</td>
<td>-FAA STC No SA01318LA-MM-Supplement 0219,8A, Rev B or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td></td>
<td>26/07/2005</td>
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<td>EASA.IM.A.S.01245</td>
<td>HARTZELL SERVICE CENTER</td>
<td>FAA STC</td>
<td>02181AK</td>
<td>CESSNA 208, CESSNA 208B</td>
<td>Installation of a Hartzell HC-B3FTN-3AF (Y7l1006CC(N)-2 three-bladed propeller and D-4897[P] assembly on the type Cessna 208, 208A and 208B in accordance with the application dated 22 April 2005 FAA STC.</td>
<td>Data package containing the following documents and additional supporting data: 1. Noise test report for SE-LZY dated 22 April 2005.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td></td>
<td>07/08/2006</td>
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<td>EASA STC No.</td>
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<td>EASA:IMA.S.01246</td>
<td>JET AVIATION AG, BASEL</td>
<td>B737-700</td>
<td>The installed CT7 Zonal Comfort System - CAB system consists of: Zonal Dryer - (behind aft cabin compartment) Humidifier 1 (FWD) - (LH Air Conditioning Mixing Bay/Humidity 2) - (RH Air Conditioning Mixing Bay/Humidity Control Box) - (Center Rack fwd Cargo Compartment-Control Panel) - (IFE Cabin)</td>
<td>As per Master Data List BADAS004-001</td>
<td>FAA STC</td>
<td>1. This STC is only valid for Serial Number 295583. It is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” Compatibility with other aircraft/engines configurations shall be determined by the installer.</td>
<td>14/07/2005</td>
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<td>EASA:IMA.S.01247</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01720LA-D LEARJET 36A LEARJET 35A LEARJET 31</td>
<td>Dual Honeywell MIST-67A Diversity Transponders for Elementary &amp; Enhanced Surveillance</td>
<td>Master Document List BAS39501075 revision F dated April 6, 2005 or later approved revision</td>
<td>FAA STC ST00961AT-D</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engines configurations shall be determined by the installer.</td>
<td>26/07/2005</td>
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<td>1001548</td>
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<td>EASA:IMA.S.01248</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03045AT-D GULFSTREAM GV-SP</td>
<td>Aircraft completion and installation of a custom cabin interior</td>
<td>In accordance with the FAA STC ST03045AT-D and Gulfstream Aerospace Corporation Index List No GC52008457 Rev C dated 27 April 2005 or later FAA approved revisions.</td>
<td>FAA STC ST00961AT-D</td>
<td>GV-SP Aircraft Serial Number 5057 Only. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engines configurations shall be determined by the installer.</td>
<td>01/07/2005</td>
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<td>EASA:IMA.S.01251</td>
<td>DECA AVIATION ENGINEERING LTD.</td>
<td>-</td>
<td>A321-211, SN 675 Interior reconfiguration 20/152 Y seating arrangement</td>
<td>1. This STC in only valid for A321 MSN 675. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engines configurations shall be determined by the installer.</td>
<td>FAA STC ST00452WI</td>
<td>1. This STC is limited to A321 MSN 675.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engines configurations shall be determined by the installer.</td>
<td>18/07/2005</td>
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<td>EASA:IMA.S.01252</td>
<td>ASSOCIATED AIR CENTER, L.P.</td>
<td>FAA ST10474SC</td>
<td>A319-133 Installation of an Executive Style Interiors... MDL 2300005091 Rev. D</td>
<td>AFM Supplement: 2300005091 Rev. D or later EASA approved and accepted revisions/AFM Supplement/Airworthiness Limitations/OMA 2300050900</td>
<td>FAA STC ST00452WI</td>
<td>1) Ref.: 2300005091 Rev. D or later EASA approved and accepted revisions/AFM Supplement/Airworthiness Limitations/OMA 2300050900</td>
<td>21/11/2005</td>
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<td>EASA:IMA.S.01254</td>
<td>DECA AVIATION ENGINEERING LTD.</td>
<td>TCCA STC G-LISA4-006/D</td>
<td>A320-232 Passenger Seat Cushion Assembly Replacement on B/E Aerospace, seat typ: D31111Txx-01 series</td>
<td>Section 1.0 of DECA Modification Summary No. MS03186, revision 1, dated January 27, 2004 or later FAA approved revision</td>
<td>FAA STC ST0061AT</td>
<td>1. This STC is limited to the A320 MSN 496.</td>
<td>18/10/2005</td>
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<td>DECA AVIATION ENGINEERING LTD.</td>
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<td>TCCA STC O-LSA05-018/D</td>
<td>A320-232</td>
<td>Reconfiguration of existing 180 passenger all economy seating arrangement</td>
<td>DECA Aircraft Engineering Modification Summary MS59917, Revision 1, dated January 20, 2005 or later EASA approved revision</td>
<td>18/10/2005</td>
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<td>TIMCO AVIATION SERVICES INC.</td>
<td>REV. 1</td>
<td>FAA STC ST0099AT-D</td>
<td>BOEING 767-300</td>
<td>Cabin Standardization in accordance with TIMCO Master Data List G37012001, revision D, dated November 30th, 2006, or later FAA approved revision.</td>
<td>This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated unless it is determined by the installer that the relationship between this change and any of those previously approved modifications will produce no adverse effect.</td>
<td>01/12/2006</td>
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<td>JAMCO AMERICA, INC.</td>
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<td>FAA STC ST01211SE</td>
<td>BOEING 767-400ER BOEING 767-300 BOEING 767-200</td>
<td>Installation of a new security flight deck door</td>
<td>JAMCO America America Master Drawing List IA-08201 MDL Revision 6, dated March 18, 2003 or later approved revision.</td>
<td>21/06/2005</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
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<td>FAA STC ST0299AT-D</td>
<td>GV-SP</td>
<td>Gulfstream Aerospace Corporation Modification FAA STC ST 0299AT-D Installation of wireless LAN system</td>
<td>Gulfstream DAS Report Number GC51309K907 dated 28/06/05. Flight Manual Supplement ref GC 11.0308/002 dated 06/06/05. - 3030309003 dated 06/06/05 for - 505 config, or later approved revisions.</td>
<td>09/06/2005</td>
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<td>BOMBARDIER AEROSPACE</td>
<td>REV. 1</td>
<td>TCCA C-LSA05-294/D</td>
<td>CL-600-2B16</td>
<td>Aircraft Completion 9 Passenger Interior Configuration. (ASN 5595)</td>
<td>Aero Consulting Services Modification Data Summary E328000 Revision N/C or later approved revision. Aero Consulting Services AFM Supplement E326070 Revision N/C or later approved revision.</td>
<td>04/04/2006</td>
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<td>BOMBARDIER INC.</td>
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<td>C-LSA05-296/ D</td>
<td>BOMBARDIER CL-600-2B16 (CL-604)</td>
<td>Aircraft Completion 11 Passenger Interior Configuration</td>
<td>Valid only for ASN 5596 Registration GV-3GM</td>
<td>07/06/2005</td>
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<td>BOMBARDIER INC.</td>
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<td>TCCA STC C-LSA05-321/D</td>
<td>BOMBARDIER CL-600-2B16 (CL-604)</td>
<td>Aircraft Completion 9 Passenger Interior Configuration</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>15/11/2005</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.01263</td>
<td>BOMBARDIER INC.</td>
<td>-</td>
<td>C-LSA06-022/ D</td>
<td>BOMBARDIER CL-600-2B16 (CL-604 Aircraft Completion, 12 Passenger Interior Configuration</td>
<td>Aero Consulting Services Ltd Modification Data Summary E267000 Revision N/C or later approved revision.</td>
<td>Restricted to ASN 5602</td>
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<td>20/01/2006</td>
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<td>EASA.IM.A.S.01264</td>
<td>BOMBARDIER INC.</td>
<td>-</td>
<td>BOMBARDIER BD-100-1A10 (CHALLENGER 604) Installation of an Executive interior on aircraft 20047 in accordance with MDL E449000</td>
<td>AFM Supplement E449000</td>
<td>1. This STC is limited to SN 20047. 2. Certification Basis: an equivalent Safety Finding has been granted in CRI D-8 (Emergency exit Marker and Locator Sign) A deviation from paragraph 25.785(b) has been granted for personal injury criteria of dynamic testing of side facing single occupancy seats...</td>
<td>07/11/2005</td>
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<td>EASA.IM.A.S.01265</td>
<td>BOMBARDIER INC.</td>
<td>REV 1</td>
<td>TCCA C-LSA06-051/D</td>
<td>BOMBARDIER BD-100-1A10 (CHALLENGER 604) Aircraft completion, 9 passengers interior configuration in accordance with Aero Consulting Services Ltd. Modification Data Summary E840000 Revision N/C</td>
<td>AFM Supplement E840000</td>
<td>1. This STC is limited to SN 20094. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>02/08/2006</td>
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<td>EASA.IM.A.S.01268</td>
<td>DUNCAN AVIATION, INC.</td>
<td>-</td>
<td>ST00315WH</td>
<td>LEARJET 60 Installation of Collins TCAS-94 i.a.w. Dok List 10045000</td>
<td>AFM 40150000</td>
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<td>EASA.IM.A.S.0135</td>
<td>AIR MEDIC L.L.C.</td>
<td>REV. 1</td>
<td>FAA STC SA076WM</td>
<td>AIR MEDIC L.L.C. Installation of Stretch Carrier in accordance with AIR MEDIC MasterDrawing List Ref: AM-2001, Revision XP, dated April 11, 2003 (or later EASA approved Revisions)</td>
<td>1. This SIC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered.</td>
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<td>EASA.IM.A.S.0136</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA7076NM</td>
<td>777-200</td>
<td>Installation of Stretcher Carrier in accordance with AIR MEDIC MasterDrawing List Ref: AM-2001, Revision ‘AE’, dated November 4, 1996 (or later EASA approved Revisions).</td>
<td>777-200</td>
<td>FAA STC SA7076NM Installation of Stretcher Carrier in accordance with AIR MEDIC MasterDrawing List Ref: AM-2001, Revision ‘AE’, dated November 4, 1996 (or later EASA approved Revisions).</td>
<td>1 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configuration shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked.</td>
<td>16/09/2004</td>
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<td>EASA.IM.A.S.0137</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA7076NM</td>
<td>737-900, 737-700, 737-800</td>
<td>Installation of Stretcher Carrier in accordance with AIR MEDIC MasterDrawing List Ref: AM-2001, Revision ‘AN’, dated April 15, 2002 (or later EASA approved Revisions).</td>
<td>737-900, 737-700, 737-800</td>
<td>FAA STC SA7076NM Installation of Stretcher Carrier in accordance with AIR MEDIC MasterDrawing List Ref: AM-2001, Revision ‘AN’, dated April 15, 2002 (or later EASA approved Revisions).</td>
<td>1 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configuration shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked.</td>
<td>16/09/2004</td>
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<td>EASA.IM.A.S.0138</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA7076NM</td>
<td>777-300</td>
<td>Installation of Stretcher Carrier in accordance with AIR MEDIC MasterDrawing List Ref: AM-2001, Revision ‘AF’, dated November 4, 1998 (or later EASA approved Revisions).</td>
<td>777-300</td>
<td>FAA STC SA7076NM Installation of Stretcher Carrier in accordance with AIR MEDIC MasterDrawing List Ref: AM-2001, Revision ‘AF’, dated November 4, 1998 (or later EASA approved Revisions).</td>
<td>1 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configuration shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or revoked.</td>
<td>16/09/2004</td>
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<td>EASA.IM.A.S.01385</td>
<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>SA05933SW</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>FAA STC SA6953SW Installation of a nose gear door ground release mechanism</td>
<td>BOMBARDIER CL-600-2B16, BOMBARDIER CL-600-2A12 (CL-600-2A12</td>
<td>FAA STC SA6953SW Installation of a nose gear door ground release mechanism</td>
<td>Arkansas Modification Center Inc. Drawing List No. 660-076 rev F dated July 22, 1988 or later approved revision.</td>
<td>25/07/2005</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>Installation of a DPI Label 1800 Membrane Switch Control System</td>
<td>Arkansas Modification Center Inc Drawing List No 600-003 rev A dated September 12 1990 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;</td>
<td>25/07/2005</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>Installation of Stretch Carrier in accordance with AR MEDIC MasterDrawing List REF : AM-2001, Revision &quot;A&quot;, dated April 15, 2002 (or later EASA approved Revisions).</td>
<td>Arkansas Modification Center Inc Drawing List No 600-101 rev A dated September 07 1990 or later approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engineconfigurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered one...</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>FAA STC SA7881SW. Installation of Sensor S67-1570 Global Positioning Antenna</td>
<td>Arkansas Modification Center Inc Drawing List No 600-105 rev A dated August 30 1990 or later approved revision</td>
<td>25/07/2005</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>FAA STC SA7881SW. Installation of a cockpit Overhead Panel Extension</td>
<td>Arkansas Modification Center Inc Drawing List No 600-101 rev A dated September 07 1990 or later approved revision</td>
<td>25/07/2005</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>FAA STC SA7872SW. Installation of a cockpit Center Pedestal/Extension</td>
<td>Arkansas Modification Center Inc Drawing List No 600-100 rev A dated August 31 1990 or later approved revision</td>
<td>25/07/2005</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>FAA STC SA7872SW. Installation of a cockpit Overhead Panel Extension</td>
<td>Arkansas Modification Center Inc Drawing List No 600-100 rev A dated August 31 1990 or later approved revision</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>FAA STC SA7872SW. Installation of an access door</td>
<td>Arkansas Modification Center Inc Drawing List No 600-104 rev A dated August 8 1990 or later approved revision</td>
<td>25/07/2005</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>FAA STC SA7872SW. Installation of an access door</td>
<td>Arkansas Modification Center Inc Drawing List No 600-104 rev A dated August 8 1990 or later approved revision</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>FAA STC SA7872SW. Installation of a Rosen Sun Visor System</td>
<td>Arkansas Modification Center Inc Drawing List No 600-103 rev A dated August 30 1990 or later approved revision</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>FAA STC SA7872SW. Installation of a Rosen Sun Visor System</td>
<td>Arkansas Modification Center Inc Drawing List No 600-103 rev A dated August 30 1990 or later approved revision</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>FAA STC SA7872SW. Installation of a Rosen Sun Visor System</td>
<td>Arkansas Modification Center Inc Drawing List No 600-103 rev A dated August 30 1990 or later approved revision</td>
<td>25/07/2005</td>
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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA ST C-52816W</td>
<td>FAA STC SA7872SW. Installation of a Rosen Sun Visor System</td>
<td>Arkansas Modification Center Inc Drawing List No 600-103 rev A dated August 30 1990 or later approved revision</td>
<td>25/07/2005</td>
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<td>EASA.IM.A.S.01400</td>
<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA STC SA9609SW</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>FAA STC SA9609SW Installation of Dual Baker M1045 Cockpit Audio Systems with a Baker M1050 Cabin Paging/Chime Amplifier</td>
<td>Arkansas Modification Center Inc Drawing Lot No 660-062 rev C dated July 15 1988 or later approved revision-AFM Supplement dated July 22 1988 is required</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>25/07/2005</td>
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<td>EASA.IM.A.S.01402</td>
<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA STC SA7160SW</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>FAA STC SA7160SW Installation of Airshree 100/200 Cabin Video System</td>
<td>Arkansas Modification Center Inc Drawing Lot No 650-040 rev C dated September 10 1990 or later approved revision</td>
<td>25/07/2005</td>
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<td>EASA.IM.A.S.01403</td>
<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>SA5830SW</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>FAA STC SA5830SW Installation of Radio Master Switch System</td>
<td>Arkansas Modification Center Inc Drawing Lot No 660-040 rev C dated September 10 1990 or later approved revision</td>
<td>25/07/2005</td>
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<td>EASA.IM.A.S.01404</td>
<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>SA7461SW</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>FAA STC SA7461SW Installation of a Dual EFIS Master Switch System</td>
<td>Arkansas Modification Center Inc Drawing Lot No 660-062 rev C dated July 15 1988 or later approved revision-AFM Supplement dated July 22 1988 is required</td>
<td>25/07/2005</td>
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<td>EASA.IM.A.S.01405</td>
<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA STC SA7464SW</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>FAA STC SA7464SW Installation of Shepherd Security System</td>
<td>Arkansas Modification Center Inc Drawing Lot No 660-079B rev B dated September 13 1990 or later approved revision</td>
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<td>EASA.IM.A.S.01408</td>
<td>BOMBARDIER AEROSPACE</td>
<td>FAA SA581NE</td>
<td>CL-600-2B16</td>
<td>FAA SA581NE Replacement of Total Temperature Sensor</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>25/07/2005</td>
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<td>EASA.IM.A.S.01411</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC SA3756WE</td>
<td>B707-100-100/-200 SERIES</td>
<td>FAA STC SA3756WE Installation of Stretcher Carrier in accordance with AIR MEDIC MasterDrawing Lot Ref: AM-2001; Revision 90, dated August 3, 1978 (or laterEASA approved Revisions).</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid until otherwise surrendered or...</td>
<td>16/02/2004</td>
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<td>ROCKWELL COLLINS, Inc.d/b/a</td>
<td>ST01142W-D</td>
<td>Modification of a single or dual Collins TDR-94 or TDR-94D Mode S Transponder System installation with CTL-02 control head to a Flight ID capable Collins TDR-94 or TDR-94D Mode S Transponder installation with CTL-02E control head</td>
<td>FAA STC</td>
<td>10015531</td>
<td>FAA STC</td>
<td>26/07/2005</td>
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<td>LEARJET Inc.</td>
<td>ST00190OE-D</td>
<td>Installation of Honeywell FMZ-2000 Flight Management System (FMS) with a Honeywell CNS/1 12 Channel GPS Sensor</td>
<td>FAA STC</td>
<td>10015533</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>26/07/2005</td>
<td>Active</td>
<td>10015533</td>
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<td>GULFSTREAM AEROSPACE SERVICES CORP.d/b/a</td>
<td>ST00640CH</td>
<td>Installation of the Honeywell Enhanced Ground Proximity Warning System</td>
<td>FAA STC</td>
<td>10015532</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>09/08/2005</td>
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<td>LEARJET Inc.</td>
<td>ST01235LA-D</td>
<td>Universal Avionics TT-3000 AERO-M Satcom System Master Drawing List No: DC 25150002 Revision E or later FAA approved revision</td>
<td>FAA STC</td>
<td>10015535</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>09/08/2005</td>
<td>Active</td>
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<td>AIR MEDIC L.L.C.</td>
<td>SA3311WE</td>
<td>Installation of Stretcher Carrier in accordance with AIR MEDIC Master Drawing List Ref: AM-2001, Revision W, dated January 28, 1977 (or later FAA approved Revisions)</td>
<td>FAA STC</td>
<td>10015536</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configuration shall be determined by the installer.</td>
<td>16/09/2004</td>
<td>Active</td>
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<td>LEARJET Inc.</td>
<td>ST01258LA-D</td>
<td>Installation of a Universal UMS-1C+ Flight Management System Upgrade from Software Control Number (SCN) 662.X to (SCN) 80X.X with Data Transfer Unit</td>
<td>FAA STC</td>
<td>10015538</td>
<td>1. Universal UMS-1C+ FMS System must be installed prior to installation of this STC. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>18/08/2005</td>
<td>Active</td>
<td>10015538</td>
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<td>EASA.IM.A.S.01421</td>
<td>LEARJET Inc.</td>
<td>FAA STD: ST00412DE-D</td>
<td>LEARJET 45</td>
<td>Installation of UNS-1C Flight Management System (Software Version 604.X)</td>
<td>FAA STC</td>
<td>LEARJET 45, Master Document List BAS/4501006 revision (-) dated April 7, 2000 or later approved revisions.AFM Supplement BAS/45041005 revision (-) dated April 07, 2000 or later approved revision.</td>
<td>1. Instructions for Continued Airworthiness BAS/4501006-IFCA or BAS/4505006-IFCA dated March 28, 2000 or later approved revision required for this installation. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”.</td>
<td>18/08/2005</td>
<td>Active</td>
<td>1011537</td>
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<tr>
<td>EASA.IM.A.S.01422</td>
<td>INFLIGHT Canada Inc.</td>
<td>REV. 1</td>
<td>TCCA STC C-LSA96-037/D</td>
<td>AIRBUS A320-232</td>
<td>SONY TRANS COM Passenger Entertainment System installation note: This STC has to be revised due to a correction to the holder of the original STC.</td>
<td>FAA STC</td>
<td>Modification Data Summary ref. A71200 revision G or later EASA approved revision</td>
<td>1. This STC is limited to the A320 MSN 496 and 5252. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>27/11/2006</td>
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<td>EASA.IM.A.S.01424</td>
<td>COMTEK Advanced Structures Ltd.</td>
<td>-</td>
<td>DORNIER D328-100</td>
<td>Replacement of trolley storage unit panels</td>
<td>FAA STC</td>
<td>- Installation Instructions, Grimes Aerospace Dec. Ref. 30-2471A, Rev. A dated 10 April 1998 or FAA STC ST500CH, dated 22 April 1998</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>18/07/2005</td>
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<td>EASA.IM.A.S.01428</td>
<td>AVIATION TECHNICAL SERVICES, Inc.</td>
<td>FAA ST00299SE-E-D</td>
<td>737-300, 737-400, 737-500</td>
<td>Modification of an 8MCU FMC Tray to accept a 4MCU FMC LRU according to FAA STC ST500CH, dated 22 April 1998</td>
<td>FAA STC</td>
<td>Modification Data Summary ref. A71200 revision G or later EASA approved revision</td>
<td>According to FAA STC ST00299SE-E-D</td>
<td>29/09/2005</td>
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<td>EASA.IM.A.S.0143</td>
<td>AIR MEDIC L.L.C.</td>
<td>FAA STC</td>
<td>B720, 720 B</td>
<td>Installation of Stretcher Carrier in accordance with AIR MEDIC Master Drawing List Ref: AM-2001, Revision ‘A’ dated January 28, 1977 (or later) EASA approved Revisions.</td>
<td>FAA STC</td>
<td>- Installation Instructions, Grimes Aerospace Doc. Ref. 30-2471A, Rev. A dated 10 April 1998 or FAA STC ST500CH, dated 22 April 1998</td>
<td>1. AFMS No. AMI-AFMS-KAIR-0429A Rev. “Original”, dated 30 September 2004 is required on-board for operation with Beech 720 and 720 B series only II. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by...</td>
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<td>STC Holder</td>
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<td>TIMCO AVIATION SERVICES INC.</td>
<td>FAA STC</td>
<td>ST03197AT</td>
<td></td>
<td>BOEING 737-800</td>
<td>Complete interior reconfiguration in accordance with TIMCO Aviation Services, Inc. M/CL report 047TL0001, revision &quot;C&quot;, dated December 07, 2005, or later EASA approved revision.</td>
<td>- M/CL report 047TL0001, revision &quot;C&quot;, dated December 07, 2005, TIMCO report 047TL0001, Revision &quot;CR&quot;, dated February 9, 2005, or later EASA approved revision.</td>
<td>1. Compatibility of this design change with other previously approved modifications must be determined by the installer. 2. This STC is approved only for the installed equipment as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>17/07/2006</td>
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<td>HEADS UP TECHNOLOGIES, Inc.</td>
<td>FAA STC</td>
<td>SA05588E-D</td>
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<td>MSN 3061 DORNIER 328-100 DORNIER 328-100</td>
<td>Installation of a Pilot and Co-Pilot Digital Air Data Unit</td>
<td>Master Data List, document No 34-06-0002-00, Revision IR, dated 23 July 2004 or later approved revision. Instructions for Continued Airworthiness - FAA STC no. ST01463SE for this modification.</td>
<td>1. Airplane Flight Manual Supplement AFMS Document 34-07-0002-00, Revision IR, dated 19 August 2004 or later approved revision. or Document 34-07-0002-00, Revision IR or later approved revision referred to in the paragraph &quot;Description&quot;.</td>
<td>15/08/2005</td>
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<td>HONEYWELL INTERNATIONAL INC.</td>
<td>FAA STC</td>
<td>SA1225O</td>
<td></td>
<td>BOEING 747</td>
<td>Honeywell International Inc. of Redmond, Washington, USA has applied for an STC to cover the installation of a CAS81 Traffic Collision Avoidance System (TCAS II) with dual TRA-67 Mode S transponders on Boeing 747-100/200/300/SP Series aircraft. The FAA has issued STC no. SA3122SO for this modification.</td>
<td>Refer to Approval Report IV/H/1TB/TPC-05.533054</td>
<td>1. Refer to Approval Report IV/H/1TB/TPC-05.533054.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>17/08/2005</td>
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<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>ST01463SE</td>
<td></td>
<td>LEARJET 31</td>
<td>Group Aircraft Approval for Reduced Vertical Separation Minimum (RVSM)</td>
<td>- Learjet Master Drawing List (MLD) BASS1015030 Revision N/C, dated April 19, 2005, or later approved revision. - Aeronautic AFM Supplement AM- AFMS-0428 Rev. A approved May 09 or later approved revision. - Instrustions for Initial &amp; Continued Airworthiness AM-015A-L-0421 Revision B dated May 5, 2005 or later.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>08/12/2005</td>
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<td>AEROSYS Inc.</td>
<td>FAA STC</td>
<td>ST01943A</td>
<td></td>
<td>MD-11F</td>
<td>Installation of the main deck cargo compartment one minute smoke detector in accordance with FAA STC ST01943A or later EASA-approved revision.</td>
<td>See item 5 of approval report (attachment)</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>EASA.IM.A.S.01440</td>
<td>AVOIDON DIVISION BUTLER NATIONAL CORP</td>
<td>35, 35A, 36, 36A</td>
<td>Installation of dual openable hatches on the belly of the fuselage.</td>
<td>Avcon Master Drawing List 1054 Revision A dated April 15 2005 or later approved revision AFM Supplement 6; Change 1 dated June 30 2005 or later approved revision.</td>
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<td>20/02/2006</td>
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<td>EASA.IM.A.S.01451</td>
<td>PRATT &amp; WHITNEY</td>
<td>CESSNA 208B CESSNA 208A CESSNA 208 BEECH C90 BEECH 300 BEECH 200 BEECH 1900</td>
<td>Installation of Altair Avionics Corporation Data Transmission Unit in accordance with instructions listed in Approved Model Data List (AML) DTU-T-225-1, Rev. - or later approved revisions. FAA STC SA001980B refers</td>
<td>Approved Model List (AML) DTU-T-225-1</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engineconfigurations shall be determined by the installer.</td>
<td>11/04/2008</td>
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<td>EASA.IM.A.S.01452</td>
<td>S-TEC CORPORATION</td>
<td>CESSNA A152 CESSNA 152</td>
<td>Installation of S-TEC System 40/50 Single and Two Axis Automatic Flight Guidance System, Model ST-628, according to Bulletin No 728, dated 3-29-96 or later approved revisions. -S-TEC Master Drawing List No 92021, dated 3-29-96, or later approved revisions. FAA/DAS Approved.</td>
<td>-S-TEC Master Drawing List No 92021, dated 3-29-96, or later approved revisions. FAA/DAS Approved. FAA/DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N 891246, dated 4-03-96 or later approved revisions is required for Cessna Models 152 and A152 for S-TEC System 40/50.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engineconfigurations shall be determined by the installer.</td>
<td>10/09/2008</td>
<td>Active</td>
<td>1015505</td>
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<tr>
<td>EASA.IM.A.S.01453</td>
<td>S-TEC CORPORATION</td>
<td>CESSNA FA-152 CESSNA F152</td>
<td>Installation of S-TEC System 40/50 Single and Two Axis Automatic Flight Guidance System, Model ST-628, according to Bulletin No 728, dated 3-29-96 or later approved revisions. -S-TEC Master Drawing List No 92021, dated 3-29-96, or later approved revisions. FAA/DAS Approved.</td>
<td>-S-TEC Master Drawing List No 92021, dated 3-29-96, or later approved revisions. FAA/DAS Approved. FAA/DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N 891247, dated 4-03-96 or later approved revisions is required for Reims Aviation S.A Models Cessna F152 and FA152 for S-TEC System.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engineconfigurations shall be determined by the installer.</td>
<td>10/09/2008</td>
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<td>EASA.IM.A.S.01454</td>
<td>S-TEC CORPORATION</td>
<td>CESSNA 210</td>
<td>Installation of S-TEC System 40/50 Single and Two Axis Automatic Flight Guidance System, Model ST-212-40/50, according to Bulletin No 312, dated 3-21-88 or later approved revisions. -S-TEC Master Drawing List No 92234, rev. D, dated 3-21-98, or later approved revisions. FAA/DAS Approved Supplements to Pilot's Operating Handbook and/or FAA Approved Airplane Flight Manual, as defined in the Limitations and Conditions section of FAA STC SA5302SW-D, latest approved revision - Instruction.</td>
<td>-S-TEC Master Drawing List No 92234, rev. D, dated 3-21-98, or later approved revisions. FAA/DAS Approved Supplements to Pilot's Operating Handbook and/or FAA Approved Airplane Flight Manual, as defined in the Limitations and Conditions section of FAA STC SA5302SW-D, latest approved revision.</td>
<td>1. As defined in the Limitations and Conditions section of FAA STC SA5302SW-D, latest approved revision. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effects.</td>
<td>06/09/2008</td>
<td>Active</td>
<td>1015507</td>
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<td>EASA.IM.A.S.01456</td>
<td>S-TEC CORPORATION</td>
<td>AERMACCHI F 260 D AERMACCHI F 260 C AERMACCHI F 260 B</td>
<td>Installation of S-TEC System 40/50 Single and Two Axis Automatic Flight Guidance Systems Model ST-235-40/50/95 FAA STC SA5242SW-D -Master Drawing List No. 92261 Rev. C dated 21.3.91</td>
<td>-Master Drawing List No. 92261 Rev. C dated 21.3.91</td>
<td>1. Only eligible on model F360B when equipped with Albaron Assemblies P/N SF260-02-05-01A (L.H.) and P/N SF260-02-05-02A (R.H.). This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer.</td>
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<td>EASA.IM.A.S.01460</td>
<td>PREMIER AIR CENTER, LLC d/b/a WEST STAR AVIATION</td>
<td>ST01884CH</td>
<td>MYSTERE FALCON 20- F5</td>
<td>Installation of a Honeywell Mark VII Enhanced Ground Proximity Warning System with Terrain Display</td>
<td>See item 5 of approval report (attachment)</td>
<td>1. EASA approved Flight Manual Supplement, Premier Air Center Document 11-327B001, Revision IR, or later EASA approved revision if required part of this STC (Configuration-012). EASA approved Flight Manual Supplement, Premier Air Center Document 11-327B004 Revision IR, or later EASA approved revision if required part of this STC (Configuration-012).</td>
<td>29/06/2006</td>
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| EASA.IM.A.S.01461 | BOMBARDIER INC. | TCCA STC Q-LSA05-030/D | BD-700-1A10 | Installation of Replacement Cockpit Printer | Maintenance Manual Supplement MMK380801 revision IR dated 11 June 2005 | 1. Installation is limited to S/N 9086
2. Note: All Portable Electronic Equipment/Devices PED, must be unplugged, turned-off and stowed prior to taxiing, take-off and landing | 12/12/2005 | Active | 10015563 |
2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description.” Compatibility with other aircraft/engine configurations shall be determined by the installer. | 23/09/2005 | Active | 10015564 |
<p>| EASA.IM.A.S.01464 | BOMBARDIER AEROSPACE | C-LSA05-212/D, ISSUE 1 | BOMBARDIER BD-100-1A10 (CHALLE) | Installation of a Satellite Telephone System Interfaced to an existing Flight Telephone System in accordance with MDL BAS68010010 Revision A dated 30 May 2005 and Compliance Report BAS6838005 dated 10/09/2005 | Maintenance Manual Supplement MMK380801 revision NC dated 15 April 2005 | 1. This STC is limited to BD-100-1A10 (Challenger 300) S/N 20013 | 05/10/2005 | Active | 10015565 |
| EASA.IM.A.S.01474 | BOMBARDIER AEROSPACE | REV 1 | TCCA STC C-LSA06-015/D ISSUE 2 | Aircraft Completion | Transport Canada STC C-LSA06-015/D Issue 2 dated September 12, 2006Aero Consulting Services Ltd Modification Data Summary E270000 Revision A or later approved revision.Aero Consulting Services Ltd AFM Supplement E270050 Revision N/C or later approved revision. | Restricted to ASN 5605 | 27/09/2006 | Active | 10015567 |
| EASA.IM.A.S.01479 | HONEYWELL INTERNATIONAL INC. | FAA STC SA0942W1-D | BEECH 1900D | Installation of the Honeywell Mark VI Enhanced Ground Proximity Warming System | Master Drawing List 159-08250-000 1 Rev. A dated 15 December 1996Aero Consulting Services Ltd AFM Supplement E270050 Revision N/C or later approved revision | AFMS 966-03875-000 is required on-board for operation with Beech 1900 D series | 28/11/2005 | Active | 10015872 |
| EASA.IM.A.S.01480 | AVIATION COMMUNICATION ANDSURVEILLANCE SYSTEMS, L.L.C. (ACIS) | FAA STC SA433HM | 757-200 | Installation and activation of TCAS and the XSS-550 Mode S Transponder (including ELIS/EHS) | Honeywell Master Data List ML407037 Revision F dated July 24 1990 or later approved revision.Honeywell AFM Supplement 040-0885 approved July 14 1990 or later approved revision. | 07/11/2005 | Active | 10015873 |</p>
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<td>EASA.IM.A.S.01484</td>
<td>SAFRAN CABIN BELLINGHAM, INC.</td>
<td>FAA STC ST003683E</td>
<td>BOEING 737-500</td>
<td>Installation of ATX Interior retrofit kit according to FAASTC ST003683E</td>
<td>- Health Tech Data Drawing List HPD-DL-57, Revision F dated 17 January 2001 - FAA STC ST003683E, Date revised 17 March 2006</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>01/02/2005</td>
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<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA STC ST069285C</td>
<td>GULFSTREAM 200</td>
<td>Installation of a Universal Unilink UL-601 DataLink Data Communication System</td>
<td>-Master Drawing List Engineering Document Number 30-110006, Revision K, dated October 2, 2000 or later approved revision -Aeronautical Flight Manual Supplement Rev 1 dated 27 September 2006, or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/10/2005</td>
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<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA STC ST069285C</td>
<td>GULFSTREAM 200</td>
<td>Installation of Hard Point Attachments and associated floorstructure</td>
<td>-Galaxy Aerospace STC data list Engineering Document Number 30-110004, revision E dated 15 November 1999 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA STC ST069285C</td>
<td>GULFSTREAM 200</td>
<td>Installation of an Artex ELT (110-406 NAV) Emergency Locator Transmitter System with NAV interface</td>
<td>-Engineering Document 30-110005, Revision K dated June 27, 2000 or later approved revision -Aeronautical Flight Manual Supplement document 30-113004 dated January 14, 2000 or later approved revision</td>
<td>1. For aircraft imported into the European Union, changes per Gulfstream Aerospace drawing CEB44231001, Rev A are required to be incorporated until Revision L or the engineering document 30-110005 has been released. 2. This STC is approved only for the product configuration as defined in the approved...</td>
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<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA STC ST097478C</td>
<td>GULFSTREAM 200</td>
<td>Installation of a NY2 Allied Signal KH8-650 HF Communication System</td>
<td>-Galaxy Aerospace STC Data List number 30-110006, revision C, dated April 18, 2000, or later approved revision -Aeronautical Flight Manual Supplement Revision 1, dated 27 September 2006, or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/10/2005</td>
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<td>EASA.IM.A.S.01494</td>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA STC ST097488C</td>
<td>GULFSTREAM 200</td>
<td>Installation of a second 77 cubic-foot oxygen bottle in nose radome area, above existing bottle</td>
<td>-Galaxy Aerospace STC data list 30-110010, revision K, dated June 26, 2003, or later approved revision -FAA Aeronautical Flight Manual Supplement GA22708M001, revision IR dated July 7, 2003 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/10/2005</td>
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<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA STC ST069219S</td>
<td>GULFSTREAM 200</td>
<td>Installation of interior shell, forward cabin, card tables, lavatory partitions and door, curtain</td>
<td>-Interior Master Drawing Document number IMD1660-012, Revision AC, dated August 28, 2002, or later approved revision -Aeronautical Flight Manual Supplement document 30-113004 dated March 12, 2001, or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>GULFSTREAM AEROSPACE Corp.</td>
<td>FAA STC: ST099238C</td>
<td>GULFSTREAM 200</td>
<td>Installation of B/E, Aerospace Aircraft Modular Products Forward/Rearward Facing Single Seat, model 2524.S46.01 ( )</td>
<td>Gulfstream STC Data List 36-1150016, Revision C, dated November 9, 2001, or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/10/2005</td>
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<td>FAA STC: ST099496C</td>
<td>GULFSTREAM 200</td>
<td>Installation of an Oxygen Filler Panel with Pressure Gauge</td>
<td>Gulfstream STC Data List 36-1150017, Revision C, dated August 22, 2001, or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/10/2005</td>
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<td>DECA AVIATION ENGINEERING Ltd.</td>
<td>TCCA STC: LSA01-426/D</td>
<td>A320-232</td>
<td>Emergency exit sign Relocation</td>
<td>DECA Drawing 7506439, Revision A, dated April 5, 2001 or later EASA approved revision</td>
<td>1) Limited to the A320 MSN 4962 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>25/10/2005</td>
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<td>S-TEC CORPORATION</td>
<td>FAA STC: SA92394AC-D</td>
<td>CESSNA F172-N</td>
<td>Installation of S-TEC System 20130 Single and tow Axis Automatic Flight Guidance Systems</td>
<td>Bulletin No 774, dated 11 June 1997 and Master drawing (techNo 921007), dated 11 June 1997 or later FAA Approved revision of the above data EASA approved Pilot’s Operating Handbooks/Aviation Flight Manual Supplement P/M EASA/R1427, dated 14 June 2005 is required for S-TEC System 20 or later EASA.</td>
<td>Also eligible on the above models when modified by 180 HP Engine conversion. EASA approval of this STC must be obtained. This STC is issued as a result of the validation of the S-TEC Corporation STC No SA92394AC-D approved by the FAA on June 26, 1997.</td>
<td>04/08/2005</td>
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<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD</td>
<td>TCCA/LSTC Q-LSA05-023/D</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Miscellaneous modifications to cabin interior and installation of Electronic Flight Bag System provisions for BD700-1A10 Global Express,.. MDL-0510002 Rev. B or later TCCA approved rev/Valitation of imported TCCA/LSTC: Q-LSA05-026/D</td>
<td>FCOM Supplement: FA955510305 Rev-A</td>
<td>1. Limited to S/N 90172. Regarding the installation of the Electronic Flight Bag System the design covers only the mounting device and its connection to the aircraft. The suitability, integrity or accuracy of the EFB hardware, software, database, display, controls and lighting, has not been assessed.</td>
<td>18/11/2005</td>
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<td>ARMSTRONG AEROSPACE Inc.dba ASTRONORDIC ARMSTRONG AEROSPACE</td>
<td>FAA ST01742CH</td>
<td>747-400F 747-400 BCF 747-400</td>
<td>Installation of Honeywell RESCU 406AF Emergency Locator Transmitter (ELT) system</td>
<td>Armstrong Aerospace master Data List Document DC104-1011-04 Revision M dated March 25, 2005 or later approved revision/armstrong Aerospace AFM Supplement Document number DC303-1011-00 Revision C approved April 25, 2005 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>25/10/2005</td>
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<td>BOMBARDIER INC.</td>
<td>TCCA LIMITED STC C-LSA05-347/D</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Installation, Aircraft Interior Completion for BD700-1A10 Global Express,.. MD5 E535000 Rev. N/C/Validation of imported TCCA Limited STC: C-LSA05-347/D</td>
<td>AFM Supplement G-FM2500013 Rev. N/C or later TCCA approved ICA G3919300011FAAMM Supplement/Avionworthiness Limitations G-TM09100089 ref. G-FM2500013 Rev. N/C</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA/STC/01504</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>REV 1</td>
<td>TCCA STC C-LSA65-272/D</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>1. Installation Aircraft Interior Completion for BD700-1A10 Global Express acc to MD5535000 Rev N/C.2. Revision 1: Modification to Interior Completion.</td>
<td>1. AMM Supplement G-FM25000132 Rev N/C or later approved revision.2. ICAO G10950001-IFCA AMM Supplement. Airworthiness Limitations G-7M0510068-4. Revision 1.1. Aero Consulting Services Ltd Modification Data Summary No E553000 Revision A dated 16 October 2008 or later approved rev.</td>
<td>1. Ref G-FM25000132 Rev N/C Limited to SN 9159.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.3. Prior...</td>
<td>26/11/2008</td>
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<td>BOMBARDIER AEROSPACE CORPORATION</td>
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<td>TCCA C-LSA65-002/D</td>
<td>BD-100-1A10</td>
<td>Cabin mod for passenger seat upgrade in accordance with Bombardier MDBA680100009</td>
<td>None</td>
<td>1. SN 2003, validation of TCCA STC C-LSA65-002/D2. Instructions for continuing airworthiness and supplemental maintenance information are provided in Airtech Aeronautical Limited's documents DMS592013. Bombardier Aerospace service bulletin No. 100-53-06 and drawingBAS68030019 must be installed prior to...</td>
<td>04/05/2005</td>
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<td>EASA/STC/01506</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST396ICH</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Flight Management System</td>
<td>- Gulfstream Aerospace Drawing List No 651-007-307-2 Revision 1 dated 11 February 2002. Gulfstream AMM Supplement Nocli AFMS-0015 Revision G dated 11 February 2002</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>12/09/2005</td>
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<td>EASA/STC/01507</td>
<td>BOMBARDIER Inc.</td>
<td>FAA STC SA556NE</td>
<td>CL-600-2B16</td>
<td>Installation of a Forward Overboard Servicing</td>
<td>I.A.E.L. Modification List ML 13545 Revision A dated January 7, 1988 or later approved revision</td>
<td>Limited to Airplane Serial Number 5059</td>
<td>25/10/2005</td>
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<td>EASA/STC/01508</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC SA556GL</td>
<td>CL-600-2B16</td>
<td>Installation of Passenger Oxygen System and Modification of Crew Oxygen System</td>
<td>- K-C Aviation Inc Engineering Drawing List 600-017 Revision M dated 17 October 1989. AMM Supplement AFMS-26 for option dated 27 August 19871. Canadian Part Numbers 600-34012-73, located at B.L.6 and 600-34012-71, located at BL.13, 73, must be replaced if any cracks are detected during the inspections required at intervals of 1000 take-off/landing cycles with this modification installed.2. Page B of K-C Aviation Instructions for Continuation, Limited documents Amtech Aeronautical Consulting Services Ltd. Information are provided in supplement maintenance airworthiness and Instructions for continuing...</td>
<td>12/09/2005</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC SA3716SW</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Installation of a bagage compartment smoke detector system</td>
<td>- K-C Aviation Inc. Drawing List No T10-049 Revision C dated 3 September 1987</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC SA4732SW</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Installation of an emergency lighting system</td>
<td>- K-C Aviation Inc. Drawing List No T10-010 Revision D</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC SA3533SW</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Installation of a Loral F-1000 Flight Data Recorder</td>
<td>- K-C Aviation Inc. Drawing List No T10-075 Revision A dated 17 June 1992 K-C Aviation Inc AFMS-244 Initial Release dated 30 June 1992</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>AIM AVIATION RENTON, INC.</td>
<td>ST015043E</td>
<td>757-200</td>
<td>Installation of a Flight Deck Galley</td>
<td>AIM Aviation Inc. STC Master Drawing List AIM. MOL-14402 Revision New dated January 25 2005 or later approved revision.</td>
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<td>03/04/2006</td>
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<td>EASA.IMA.S.01537</td>
<td>TIMCO AVIATION SERVICES INC.</td>
<td>FAA STC ST03107AT-D</td>
<td>BOEING 767-300</td>
<td>Interior Reconfiguration</td>
<td>TIMCO Master Data List 04T110001 Revision IR dated December 08 2005 or later approved revision.</td>
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<td>EASA.IMA.S.01538</td>
<td>ROCKWELL COLLINS AEROSPACEAND ELECTRONICS, Inc.</td>
<td>ST01205NY</td>
<td>BOMBAIDER CL-600-2B16</td>
<td>Installation of Head Up Guidance System (HGS) Model 2100 (Phase 2)</td>
<td>STC Master Drawing List No. 9701-1085 Revision N or later approved revision.AFP Supplement 9701-1087 Revision D or later approved revision.</td>
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<td>EASA.IMA.S.01539</td>
<td>BOMBAIDER AEROSPACE</td>
<td>TCCA STC C-LSA05-314D</td>
<td>BOMBAIDER CL-600-2B16</td>
<td>Aircraft Completion 11 Passenger/Interior Configuration</td>
<td>Aero Consulting Services Modification Data Summary E265000 Revision N/C or later approved revision.Aero Consulting Services AFM Supplement E265000 Revision N/C or later approved revision.</td>
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<td>1. Restricted to Aircraft Serial Number 5599.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.IMA.S.01540</td>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SA285SH-D</td>
<td>CESSNA T337H-SP CESSNA T337H CESSNA T337F CESSNA T337E CESSNA T337D CESSNA P337H CESSNA 337H CESSNA 337G CESSNA 337F CESSNA 337E CESSNA 337D</td>
<td>Installation of S-TEC System 60 Two Axis Automatic Flight Guidance System with Optional Autotrim System, Model ST-314, according to SAE8459, dated 4-06-90 or later approved revision.</td>
<td>-S-TEC Master Drawing List No:80503, dated 3-14-90 or later approved revision.FAA/DAS Approved Supplemental Flight Manual, Part No 88845, dated 4-06-90 is required for Cessna Models 337D, T337D, 337E, T337F, 337F, T337G, T337G, 337H,P337H, T337H, and T337H-SP when modified per FAA STC SA1627WE.</td>
<td>1. Eligible when above models are modified per FAA STC SA1627WE (Robertson STOL Conversion).2. Eligible when the above models are modified per FAA STC SA937CE (Horton STOL Conversion).3. It is the operator’s responsibility to ensure the above mentioned Horton &amp; Robertson STOL Conversion.4. This STC is approved only for the p...</td>
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<td>EASA.IMA.S.01541</td>
<td>S-TEC CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC SA210SH-D</td>
<td>REIMS AVIATION FT-337HP REIMS AVIATION FT-337P REIMS AVIATION FT-337F REIMS AVIATION FT-337 REIMS AVIATION F-337H REIMS AVIATION F-337G REIMS AVIATION F-337F REIMS AVIATION F-337</td>
<td>Installation of S-TEC System 60 Two Axis Automatic Flight Guidance System with Optional Autotrim System, Model ST-314, according to SAE8459, dated 4-06-90 or later approved revision.</td>
<td>1. S-TEC Master Drawing List No:80503, dated 3-14-90, or later approved revisions.2. FAA/DAS Approved Supplemental Flight Manual, Part No 88845, dated 4-06-90 is required for Reims Aviation S.A. Models P337E, FT337E, FT337F, FT337G, FT337GP, FT337HP and FT337HP when modified per FAA STC SA1627WE.</td>
<td>1. Eligible when the above models are modified per FAA STC SA1627WE (Robertson STOL Conversion).2. Eligible when the above models are modified per FAA STC SA937CE (Horton STOL Conversion).3. It is the operator’s responsibility to ensure the above mentioned Horton and Robertson STOL Conversion...</td>
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<td>MYSTERE FALCON 50</td>
<td>GPWS upgrade to EGPWS</td>
<td>Master Data List Doc. No. BEHHS003-001 Revision A or later approved revisions.AFP Supplement Doc. No. BEHHS003-201-01 Revision IR or later approved revisions.ICA Doc. No. BEHHS003-150-01 Revision IR or later revisions.</td>
<td>This STC is limited to Falcon 50 SN 1/2</td>
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<td>Limitations/Conditions</td>
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<td>HYDRO-AIRE AEROSPACE CORP.</td>
<td>FAA ST01544-AT</td>
<td>KING AIR 200 BEECH 200, SUPER KING AIR</td>
<td>Installation of Power Brake / Slot Control System</td>
<td>Model 200 Hydro-Aire Aircraft Drawings No. 13-001 Revision C or later approved revision Model B200 Hydro-Aire Aircraft Drawings No. 13-001-100, Revision C or later approved revision.</td>
<td>1. Models 200 Hydro-Aire Aircraft Fatality Manual Supplement dated on 17 May 1977 or later approved revision. 2. Models B200 Hydro-Aire Aircraft Fatality Manual Supplement No. R13-001-100-06 dated 12 May 2006 or later approved revision. 3. This STC is approved only for the product configuration as def...</td>
<td>02/20/2020</td>
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<td>EASA.IMA.S.01546</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>ST01824CH-D</td>
<td>MD-88 GULFSTREAM ASTRAS DX, GULFSTREAM 100</td>
<td>Installation of the Honeywell MK-V ECS/PWS, with windshear warning and terrain display on the Collins Pro Line 4 EFIS</td>
<td>GULFSTREAM DAS Index List ref GC51103A06 and report No. GC51103B000 dated 6/10/2005 or later approved revisions/FMS GC51103M000/FAA STC ref ST01824CH-D.</td>
<td>FAA STC No. ST01824CH-D. Master Drawing List GA113498B001 Rev. B, dated March 30, 2005 or later EASA approved Revisions- Airplane Flight Manual Supplement Report No. GA113498001, Rev. dated 09-5-05 or later EASA approved Revisions- Instruction for Continued Airworthiness Report No. GA1134...</td>
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<td>EASA.IMA.S.01550</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST015504A</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of a Honeywell Satellite TV antenna to provide television reception to passenger cabin TV monitors</td>
<td>Gulfstream DAS Index List ref GC5115070029 and report No GC511507X000 dated 5/1/2005 or later approved revisions/FAA STC ref ST015504A-D.</td>
<td>Gulfstream DAS Index List ref GC5115070029 and report No GC511507X000 dated 5/1/2005 or later approved revisions/FAA STC ref ST015504A-D.</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA ST015514A</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of a Conduct in the vertical stabiliser to accommodate wiring for the STACOM and SAT TV antennas</td>
<td>Gulfstream DAS Index List ref GC512498B000 and report No GC512498K000 dated 9/1/2005 or later approved revisions/FAA STC ref ST015514A-D.</td>
<td>Gulfstream DAS Index List ref GC512498B000 and report No GC512498K000 dated 9/1/2005 or later approved revisions/FAA STC ref ST015514A-D.</td>
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<td>EASA.IMA.S.01553</td>
<td>LIFEPORT, LLC</td>
<td>FAA ST015534A</td>
<td>CESSNA 560 XL</td>
<td>Installation of a Patient Loading and Utility System (PLUS) in accordance with FAA STC ST01780LA or later EASA approved revision Note: Revision 1 issued to correct a type in the Original STC number.</td>
<td>Master Drawing List, Nr. MDL-777</td>
<td>Master Drawing List, Nr. MDL-777</td>
<td>Related AFM Supplement ESR-777-11, Revision A, dated June 24, 2005, or later FAA approved revision</td>
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<td>EASA.IMA.S.01554</td>
<td>ROCKWELL COLLING, Inc.</td>
<td>FAA ST015540A</td>
<td>B757-200</td>
<td>Installation of Trans Com Video Projection System</td>
<td>Trans Com Index List 062-0128 Revision 2, dated March 10 1999 or later approved revision</td>
<td>Trans Com Index List 062-0128 Revision 2, dated March 10 1999 or later approved revision</td>
<td>FAA STC ref ST015540A.</td>
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<td>EASA.IMA.S.01555</td>
<td>LEARJET Inc.</td>
<td>FAA ST015554A</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Pilot and Co-Pilot Footrests for B700-1A10 Global Expressac:::</td>
<td>FAA STC ref ST015554A.</td>
<td>FAA STC ref ST015554A.</td>
<td>Validation of imported FAA STC ref ST015554A.</td>
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### List of EASA Supplemental Type Certificates

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<th>EASA.IM.A.S.01569</th>
<th>GULFSTREAM SERVICES CORPORATION</th>
<th>FAA STC ST0915AT-D</th>
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<th>Lightning Sensor System Installation</th>
<th>FAA STC ST0915AT-D AFMS GC51 225M000</th>
<th>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</th>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03134AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Aircraft Completion STC</td>
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<td>EASA.IM.A.S.01571</td>
<td>ANDRA INTERNATIONAL LLC</td>
<td>ST01748A</td>
<td>757-200</td>
<td>Installation of a Main Deck Cargo Handling System</td>
<td>Anca International LLC Controlled Drawing List DL8388 Revision A dated March 5, 2004 or later approved revision</td>
<td>23/12/2005</td>
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<td>EASA.IM.A.S.01573</td>
<td>TRANSAIRCO S.A.</td>
<td>-</td>
<td>FALCON 20</td>
<td>Incorporation of Mode S EHS capability</td>
<td>Modification TSA F20-34.80.01AFMS F20-34.80.01 AFMS</td>
<td>1. Limited to Falcon 20 S/N 237, 257, 307, 341, 389, 402. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>TACA INTERNATIONAL AERLINES</td>
<td>- AIRBUS A320-233</td>
<td>Automatic ELT Provisions installation and activation</td>
<td>Grupo TACA Engineering Order A320MO25-099/99</td>
<td>Revision datedJuly 31, 2000</td>
<td>This STC is limited to A320 MSN 661</td>
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<td>TACA INTERNATIONAL AERLINES</td>
<td>- AIRBUS A320-233</td>
<td>Automatic ELT Provisions installation and activation</td>
<td>Grupo TACA Engineering Order A320MO25-099/99, Revision 3, dated June 7, 2001</td>
<td>This STC is limited to A320-233 MSN 558</td>
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<td>TACA INTERNATIONAL AERLINES</td>
<td>A320-233</td>
<td>Equipment and furnishing – delivery configuration of passenger cabin - 180 Pax</td>
<td>Taca Engineering Order A320MO25-12205, Revision 0, dated August 31, 2005.</td>
<td>This STC is limited to the A320-233 MSN 561</td>
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<td>TDG AEROSPACE INC</td>
<td>FAA STC ST042NM</td>
<td>MD-11 DC-9-87 (MD-87) DC-9-83 (MD-83) DC-9-83 (CF6) DC-9-81 (MD-81)</td>
<td>Installation of NOFOSS Overwing Anti-Ice System</td>
<td>TDG Aerospace Drawing List No E50-104, Rev U dated 31 October 2002</td>
<td>1. Flight Manual Supplement TDG Doc No E50-096 Rev 0 dated 27 September or later approved revision. Instructions for Continued Aeronautical TDG Doc E50-108 Revision P dated 11 February 2003. This STC is approved only for the product configuration as defined in the approved design data referred to...</td>
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<td>JET AVIATION AG</td>
<td>REV. 2</td>
<td>BOMBARDIER CL-600-2B16 BOMBARDIER CL-600-2A12 (CL-601 BOMBARDIER CL-600-1A11 (CL-600</td>
<td>Activation of RAAS</td>
<td>Master Data List No. BCL613466-001, Rev. D of 12-01-2009 or later approved revision AFM Supplement B2. BCL613466-201-01, Rev. A of 12-01-2009 or later revision.</td>
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<td>FAA STC ST01580LA-D</td>
<td>GULFSTREAM ST01580LA-D</td>
<td>GULFSTREAM G550</td>
<td>Installation of Executive Cabin Interior and other Completion Related Modifications</td>
<td>- FAA STC No ST01580LA-D-GAC Index List GC52 000491, Rev D Jan 10, 2006 or later approved revisions-FAFMG G52009/AFM 007, Rev 0, Jan 11, 2006 or later approved revisions</td>
<td>G550 S/N 5078 only</td>
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<td>ST01581LA-D</td>
<td>GULFSTREAM ST01581LA-D</td>
<td>GULFSTREAM G550</td>
<td>Installation of an Motorola Satellite Telephone (Iridium) Antennas</td>
<td>- FAA STC ST01581LA-D-GAC Index List GC5210930000, Rev D, Dec 02, 2003 or later approved revisions</td>
<td>GV-SP (550)</td>
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<td>TIMCO AVIATION SERVICES INC.</td>
<td>FAA STC SA04-113</td>
<td>TIMCO AVIATION SERVICES INC.</td>
<td>TC040-1A10</td>
<td>Installation, Completion of custom Aircraft Interior for BD100-1A10 Global Express.</td>
<td>- FAA STC SA040113 Rev 0 or later TCCA approved rev/ICA GC31450000, /FAA/AAMM Supplement G-TM05140001</td>
<td>1. Ref: GM95000401 Rev. INC. Limited to S/N 51432. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC SA02444K</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>BD-700-1A10</td>
<td>Installation of Rockwell Collins Traffic Alert and Collision Avoidance System (TCAS II) in Raytheon HS 125 Series 700A Aircraft</td>
<td>- Elliott Aviation Master Data List 2004-STDL, Rev 0, dated December 10, 2002 or later approved revisions.</td>
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<td>AIRGLAS, Inc.</td>
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<td>AIRGLAS, Inc.</td>
<td>FAA STC SA02444K</td>
<td>Fabrication and installation of Airglas Inc Model LH4000 hydraulically retractable wheel skids.</td>
<td>- FAA STC SA040113 Rev 0 or later approved revisions-FAFMG G52009/AFM 007, Rev 0, FAA Approval dated December 15, 2002, or later approved revisions-Instructions for Continued Airworthiness, Chap...</td>
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<td>TIMCO AVIATION SERVICES INC.</td>
<td>FAA STC ST01592AT-D</td>
<td>TIMCO AVIATION SERVICES INC.</td>
<td>FAA STC ST01592AT-D</td>
<td>Installation of Flight Data Recorder Expanded Parameters with a Plessey FDAU</td>
<td>Master Data List 991073701 Rev. L or later approved revision</td>
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<td>TIMCO AVIATION SERVICES INC.</td>
<td>FAA STD ST01315AT-D</td>
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<td>BOEING 767-300</td>
<td>Installation of Rockwell Collins ACARS CMU-900</td>
<td>TIMCO Master Data List 0ST036001 Revision IR dated November 21, 2005 or later approved revision</td>
<td>FAA STC ST03316AT</td>
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<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC ST0069WI</td>
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<td>BOMBARDIER CL-600-2B16, BOMBARDIER CL-600-2A12 (CL-401), BOMBARDIER CL-600-1A11 (CL-600)</td>
<td>Installation of Honeywell TCZ-910 TCAS II &amp; XS-852 Mode S Transponder System</td>
<td>1) Duncan Aviation Inc. Document List No. 10040700 Revision G dated December 3, 2002) Duncan Aviation Inc. AFM Supplement Document No. 40134800 Revision B &amp; 40141900 Revision B; or later FAA approved revisions. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>BOMBARDIER INC.</td>
<td>TCCA STC C-LSA06-014D</td>
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<td>BOMBARDIER BD-700-1A11</td>
<td>Installation of 13 Passenger interior completion for BD700-1A11</td>
<td>-</td>
<td>1. Installation of limited to S/N 91682. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engines configurations shall be determined by the installer.</td>
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<td>BOMBARDIER INC.</td>
<td>TCCA STC NO C-LSA06-382D</td>
<td>-</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Aircraft Completion 13 Passenger Interior Configurationalization of TCCA Limited STC C-LSA06-382D to S/N 9170MDL: E810000 Rev A or later approved revisions.</td>
<td>FMS: G-FM2500138 Rev A or later approved revisions/FAA CM1G91700001-IFCA</td>
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<td>LIFEPORT, INC.</td>
<td>FAA STC SA00635SE</td>
<td>-</td>
<td>208-208A-208B</td>
<td>Installation of a Patient Loading and Utility System (PLUS) in accordance with FAA STC SA00635SE or later EASA approved revision</td>
<td>See item 5 of approval report (attachment)</td>
<td>1. Airplane Flight Manual Supplement EUR-071-11, Revision A; dated June 1, 2002, or later FAA approved revision.</td>
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<td>16/02/2006</td>
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<td>EASA-IM.A.S.01620</td>
<td>GKN AEROSPACE CHEM-TRONICS Inc.</td>
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<td>AIRBUS A321, AIRBUS A320, AIRBUS A319, AIRBUS A318</td>
<td>Replacement CFM56-5B Inlet Barrel Lower Panel Assembly (Part Number 030-13001-1)</td>
<td>Master Drawing List Ref. ML0030-13014 Revision D dated April 20, 2005 or later approved revision</td>
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<td>AERO CONSULTING SERVICES LTD.</td>
<td>REV 1</td>
<td>TCCA STC C-LSA05-303/D ISSUE 2</td>
<td>BOMBARDIER CL-600-2B16</td>
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<td>- Aero Consulting Services Modification Data Summary E59800 Revision B or later approved revision - Midcoast Aviation AFM Supplement 34-24/M005 Revision N/C or later approved revision - Midcoast Aviation AFM Supplement 34-26/M017 Revision N/C or later approved revision.</td>
<td>Restricted to Aircraft Serial Number 5610</td>
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<td>EASA.IM.A.S.01625</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td></td>
<td>FAA STC SA0061NM</td>
<td>MD-88 DC-9-37 (MD-87) DC-9-83 (MD-83) DC-9-83 (MD-83) DC-9-81 (MD-81)</td>
<td>Installation of Honeywell Traffic Alert &amp; Collision Avoidance System (TCAS II)</td>
<td>Honeywell Master Data List No ML 40741700 Revision A dated March 21, 1991 or later approved revision.Honeywell AFM supplement TCAS-APPS-0248 Revision A dated October 18, 2000 or later approved revision.</td>
<td>30/01/2006</td>
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<td>EASA.IM.A.S.01626</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
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<td>FAA STC SA541NM</td>
<td>BOEING 767-300 BOEING 767-200</td>
<td>Installation of Honeywell Traffic Alert &amp; Collision Avoidance System (TCAS)</td>
<td>Honeywell Master Data List No ML 4076520 Revision N/C dated July 24, 1992 or later approved revision.</td>
<td>1. TCAS Airplane Flight Manual Supplement (Change 7) TCAS-APPS-0294, Rev A. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft/engine configurations shall be determined by the...</td>
<td>12/12/2005</td>
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<td>EASA.IM.A.S.01627</td>
<td>ITOCHU AIRLEASE, INC.</td>
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<td>FAA ST104366C</td>
<td>BOEING 767-300</td>
<td>Reconfiguration of an existing three seating class interior with 199 passengers to a two seating class interior (economy &amp; business seats) with 232 passengers.</td>
<td>Future Jet Aerodesign LLC Master Data List 1610000 Revision 7 dated July 7, 2005 or later approved revision AFM Supplement Document number 1610000-AFMS Rev IR dated July 15, 2005 or later approved revision.</td>
<td>02/11/2005</td>
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<td>EASA.IM.A.S.01628</td>
<td>AERO CONSULTING SERVICES LTD.</td>
<td>REV 1</td>
<td>TRANSPORT CANADA STC C-LSA05-2</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Aircraft Completion 12 Passenger Interior Configuration</td>
<td>- Aero Consulting Services Modification Data Summary E59800 Revision B or later approved revision - Midcoast Aviation AFM Supplement 34-24/M005 Revision N/C and 34-26/M017 Revision N/C or later approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer. 2. Restricted to Aircraft Serial Number 5609.</td>
<td>25/07/2006</td>
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<td>S-TEC CORPORATION</td>
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<td>FAA STC SA4005SW-D</td>
<td>TB20, TB21</td>
<td>Installation of S-TEC System 5S/55X Two Axis Automatic Flight Guidance System, Model ST-646</td>
<td>Bulletin No 546 Revision 3, dated 23 August 2001 and Masterdrawinginho 92/895, Revision C, dated 23 August 2001 or later approved revision of the above data (14 Volt system) ,EASA approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement/ Not EASA81094 dated 28 November 2005 is re...</td>
<td>1. This STC is issued as result of the validation of the S-TEC Corporation STC no SA4005SW-D approved by the FAA on August 23, 2001. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft...</td>
<td>12/12/2005</td>
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### List of EASA Supplemental Type Certificates

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<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>F A A S T C D T 0 0 1 5 6 6 5 7</td>
<td>HAWKER 700A AND 700B</td>
<td>System 10015665</td>
<td>In accordance with Master Drawing List MDL RASF02101, Revision 2, dated January 8, 2003 or later approved revision. FAA approved Airplane Flight Management System. Document Numbers FMS-RASF02101A and if applicable Document Number FMS-RASF0210B, dated December 18, 2002 if FMS-RASF02101C, dated March 21, 2003...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>19/12/2005</td>
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<td>WIPIRE INC.</td>
<td>F A A S T C D T 0 0 1 5 6 6 5 7</td>
<td>CESSNA 172 SERIES</td>
<td>System 10015665</td>
<td>FAA STC SA0674CH Installation of Wipaire Model 2110 Amphibian/Seaplane floats on Cessna Reims Aviation 172F/172 Models as defined in the AML.</td>
<td>FAA STC SA0674CH Installation of Wipaire Model 2110 Amphibian/Seaplane floats on Cessna Reims Aviation 172F/172 Models as defined in the AML.</td>
<td>21/06/2006</td>
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<td>10115667</td>
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<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td>F A A S T C D T 0 0 1 5 6 6 5 7</td>
<td>BOEING 757-300</td>
<td>System 10015665</td>
<td>Installation of an In-Flight Cabin Video Surveillance System</td>
<td>Installation of an In-Flight Cabin Video Surveillance System</td>
<td>16/03/2006</td>
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<td>JET AVIATION AG, BASEL</td>
<td>F A A S T C D T 0 0 1 5 6 6 5 7</td>
<td>FALCON 900</td>
<td>System 10015665</td>
<td>1. FAA Supplemental Type Certificate SA332GL is required for models 172M, 172N and 172P when installed in accordance with Wipaire Drawing 10D1-601. 2. FAA Supplemental Type Certificate SA2890AT is required for models 172, 172J, 172M, 172N and 172P when installed in accordance with Wipaire Drawing 10D1-601...</td>
<td>1. FAA Supplemental Type Certificate SA332GL is required for models 172M, 172N and 172P when installed in accordance with Wipaire Drawing 10D1-601. 2. FAA Supplemental Type Certificate SA2890AT is required for models 172, 172J, 172M, 172N and 172P when installed in accordance with Wipaire Drawing 10D1-601...</td>
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### List of EASA Supplemental Type Certificates

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<td>RAISBECK ENGINEERING Inc.</td>
<td>LEARJET 36A LEARJET 36</td>
<td>Installation of drag reduction system (including outboard trailing edge extension, leading edge vortilons &amp; contoured flap). Installation of drag reduction system (including outboard trailing edge extension, leading edge vortilons &amp; contoured flap).</td>
<td>FAA STC 10015666</td>
<td>1. Raisbeck Engineering Drawing Number 205L-4700 Revision 3 dated March 4 2005 or later approved revision. 2. Raisbeck Engineering AFM Supplement 204L-5780 Revision IR dated March 4 2005 or later approved revision.</td>
<td>21/02/2006</td>
<td>Active</td>
<td>10115670</td>
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<tr>
<td>TACA INTERNATIONAL AERONAUTICAL TECHNICAL AND TRAINING SERVICES(TTS)</td>
<td>AIRBUS A320-233</td>
<td>Equipment and furnishing - Delivery configuration of passenger cabin - 180 pax</td>
<td>FAA STC 10015668</td>
<td>Taca Engineering Order A320/025-12105, Revision 0, dated August 31, 2005. This STC is Limited to the A320 MSN 558.</td>
<td>26/10/2005</td>
<td>Active</td>
<td>10115657</td>
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<tr>
<td>HARTZELL PROPELLER INC.</td>
<td>DIAMOND DA-40</td>
<td>Installation of Hartzell Two blade propeller HC-2CYR-1BF/F7495S with STC SA01850CH in the Diamond DA40 aeroplane</td>
<td>FAA STC 10015669</td>
<td>Hartell engineering report no. 30B3Installation Instruction DA40-NST- EASAFlight Manual Supplement DA40-AFMS- EASA.</td>
<td>30/09/2005</td>
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<td>S-TEC CORPORATION</td>
<td>CESSNA 172S CESSNA 172R</td>
<td>Installation of S-TEC System 30 ALT configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>FAA STC 10015670</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>10/09/2008</td>
<td>Active</td>
<td>10115659</td>
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<td>AERO CONSULTING SERVICES LTD.</td>
<td>CL-600-2B16</td>
<td>Aircraft Completion 12 Passenger Interior Configuration</td>
<td>TCCA STC C-LSA05-317/D ISSUE 2</td>
<td>1. Restricted to ASN 5620 only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>10/08/2006</td>
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<td>10115670</td>
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<td>AERO CONSULTING SERVICES LTD.</td>
<td>CL-600-2B16 (CL-604)</td>
<td>Aircraft Completion 12 Passenger Interior Configuration</td>
<td>TCCA C-LSA05-315/D ISSUE 2</td>
<td>1. Restricted to ASN 5612 only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>11/06/2006</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC</td>
<td>ST03142AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Aircraft Completion STC</td>
<td>FAA STC Number ST03142AT-D</td>
<td>1. Limited to SN 50812. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircrafthng configurations shall be determined by the installer.</td>
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<td>BOMBARDIER AEROSPACE</td>
<td>REV 1</td>
<td>TCCA STC C-LSA06-025/D</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Aircraft Completion, 10 Passenger Interior.</td>
<td>Transport Canada STC C-LSA06-025/D Issue 2 dated August 31, 2006 or later approved revision. Aero Consulting Services Modification Data Summary E689000 Revision A or later approved revision. Aero Consulting Services AFM Supplement E689000 Revision NC or later approved revision.</td>
<td>1. Restricted to ASN 5613.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.IM.A.S.01656</td>
<td>BOMBARDIER AEROSPACE</td>
<td>FAA STC</td>
<td>ST03065AT</td>
<td>LEARJET 31A</td>
<td>Installation of a Single Universal UNS-1C or UNS-1E Flight Management System</td>
<td>Bombardier Aerospace Master Document List BAS3107200 Revision B dated February 22, 2005 or later approved revision. Bombardier Aerospace AFM Supplement ref. BAS31047000 Revision NC dated March 14, 2005 or later approved revision.</td>
<td>04/01/2006</td>
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<td>EASA.IM.A.S.01658</td>
<td>BOMBARDIER AEROSPACE</td>
<td>REV 1</td>
<td>TCCA STC C-LSA05-024/D ISSUE 2</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Aircraft Completion, 10 Passenger Interior Configuration</td>
<td>Transport Canada STC C-LSA05-024/D Issue 2 dated August 3, 2006. Aero Consulting Services Modification Data Summary E688000 Revision NC or later approved revision. Aero Consulting Services AFM Supplement E688090 Revision NC or later approved revision.</td>
<td>1. Restricted to Aircraft Serial Number 5611.3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>AERO CONSULTING SERVICES LTD.</td>
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<td>TCCA STC C-LSA05-028/D</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Aircraft Completion 12 Passenger Interior Configuration</td>
<td>Transport Canada Limited STC C-LSA05-328/D Issue 2 dated October 02, 2006 Aero Consulting Services Modification Data Summary E640000 Revision A or later approved revision. Aero Consulting Services AFM Supplement 34-24M005 Revision NC or later approved revision. Midcoast Aviation Airplane Flight Supplement 34-24M005 Revision NC or later approved revision.</td>
<td>1. Limited to SN 56172. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>ST01841CH</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>ST01841CH</td>
<td>Installation of GPS Antenna Provisions B747 Series in accordance with ST01841CH</td>
<td>Data package containing the following documents, and additional supporting data: 1. Drawing 100-91347-1-NC Structural Provisions Installation GPS Antenna B747 Series: 2. Drawing 100-91347-2-NC Structural Provisions Installation GPS Antenna Right</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer. The Boeing 747SR is not included in the EASA STC.</td>
<td>31/07/2006</td>
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<td>ST01877CH</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>ST01877CH</td>
<td>Installation of a Honeywell Ground proximity Warning System in accordance with FAA STC ST01877CH</td>
<td>Data package containing the following documents, and additional supporting data: 1. Certification Plan L1011: Installation of Honeywell International EGPS with Terrain Display on Lockheed L1011 Series Aircraft (configuration 1 upgrade), Electronic Cable Specialists Document ECS29489, Revisions N...</td>
<td>1. Airplane Flight Manual Supplement, Electronic Cable Specialists Doc. No. ECS-203895, Revision A, approved September 23, 2005, or later FAA approved revisions, are required on board the aircraft.2. FAA approved revisions, are required on board the aircraft.3. FAA Flight Manual Supplement, Electronic Cable Specialists Doc. No. ECS-203895, Revision N/C, approved September 23, 2005, or later FAA approved revisions, are required on board the aircraft.4. FAA Flight Manual Supplement, Electronic Cable Specialists Doc. No. ECS-203895, Revision N/C, approved September 23, 2005, or later FAA approved revisions, are required on board the aircraft.</td>
<td>02/06/2006</td>
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<td>ST01923CH</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>ST01923CH</td>
<td>Installation of a Honeywell Enhanced Ground Proximity Warning System in accordance with ST01923CH</td>
<td>Data package containing the following documents, and additional supporting data: 1. Certification Plan L1011: Installation of Honeywell International EGPS with Terrain Display on Lockheed L1011 Series Aircraft (configuration 1 upgrade), Electronic Cable Specialists Document ECS29489, Revisions N...</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.2. FAA approved revisions, are required on board the aircraft.3. FAA Flight Manual Supplement, Electronic Cable Specialists Doc. No. ECS-203895, Revision A, approved September 23, 2005, or later FAA approved revisions, are required on board the aircraft.4. FAA Flight Manual Supplement, Electronic Cable Specialists Doc. No. ECS-203895, Revision N/C, approved September 23, 2005, or later FAA approved revisions, are required on board the aircraft.5. FAA Flight Manual Supplement, Electronic Cable Specialists Doc. No. ECS-203895, Revision N/C, approved September 23, 2005, or later FAA approved revisions, are required on board the aircraft.</td>
<td>24/01/2006</td>
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<td>ST01887CH</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>ST01887CH</td>
<td>Installation of GPS antenna provisions in accordance with FAA STC ST01887CH</td>
<td>Certification for Reduced Vertical Separation Minimum (RVSM) operational capability and installation of Honeywell A2-252 Air Data Computer and A2-250 Altimeter with optional BA-250 Altitude Indicator and Al-800 Altitude Preselector.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.2. Certification basis: FAA TCDS A14SW for MO-30; FAA TCDS A16SW...</td>
<td>13/03/2006</td>
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<td>FAA STC SA011669l-D</td>
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<td>Certification basis</td>
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<td>EASA.IM.A.S.01672</td>
<td>AMSAFE, INC. d/b/a AMSAFE AVIATION</td>
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<td>AVIATION AIRCRAFT A-1/1</td>
<td>installation of Am-Safe Avionics Inflatable Restraint System on pilot and co-pilot seat (FAA STC SA01701A/1 issued December 06, 2004)</td>
<td>Certification basis</td>
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<td>GULFSTREAM AEROSPACE SERVICES CORP. d/b/a GULFSTREAM AEROSPACE CORPORATION</td>
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<td>Certification basis</td>
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<td>ROCKWELL COLLINS, Inc. d/b/a AIR TRANSPORT SYSTEMS</td>
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<td>STO281AT</td>
<td>BOEING 737-700</td>
<td>installation of a Rockwell Collins CMU-900 Communications Management Unit, VHF-200 VDL Mode 2 Radio and the APM-900 Airplane Personality Module to provide Controller Pilot Data Link Communications (CPDLC) Baseline 1 services</td>
<td>Certification basis</td>
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<td>EASA.IM.A.S.01681</td>
<td>ELLIOTT AVIATION</td>
<td>FAA STC ST01486CH</td>
<td>BEECH MJ-300-10 BEECH MJ-300 BEECH 400</td>
<td>Installation of a Honeywell Mark VII/EVIII EGPWS</td>
<td>Certification basis</td>
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List of EASA Supplemental Type Certificates

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## List of EASA Supplemental Type Certificates

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<tr>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC 10015701</td>
<td>01</td>
<td>ST0347AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of Airshow 4000 (Moving Map and In-Flight Information)</td>
<td>Gulfstream Aerospace Corporation Index List GC41 60468001, Revision D, dated March 14, 2005 or later approved revisions</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer.</td>
<td>08/02/2006</td>
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<td>FAA STC 10015694</td>
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<td>ST0347AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of Interior Emergency Lights</td>
<td>Gulfstream Aerospace Corporation Index List GC41 48886000, Revision E, dated March 14, 2005 or later approved revisions</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer.</td>
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<td>GULFSTREAM GIV-X</td>
<td>Installation of External Camera System (Securaplane)</td>
<td>Gulfstream Aerospace Corporation Index List GC41 45880000, Revision E, dated March 31, 2005 or later approved revisions</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer.</td>
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<td>ST0347AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of Passenger Oxygen System</td>
<td>Gulfstream Aerospace Corporation Index List GC41 45380000, Revision B, dated March 07, 2005 or later approved revisions</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer.</td>
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<td>ST0347AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of Cabin Window Shades</td>
<td>Gulfstream Aerospace Corporation Index List GC41 50880000, Revision C, dated March 14, 2005 or later approved revisions</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer.</td>
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<td>TRANSARCO S.A.</td>
<td>FAA STC 10015697</td>
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<td>ST0347AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Structural Installation of Honeywell KA-815 TCAS I Antennas</td>
<td>Engineering order TSA PC12-24,46O2EO</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engineconfigurations shall be determined by the installer.</td>
<td>09/03/2006</td>
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<td>ELECTRONIC CABLES SPECIALISTS, Inc.</td>
<td>FAA STC 10015698</td>
<td>01</td>
<td>ST0347AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of elementary and enhanced surveillance Mode-S Transponders in MD-11, MD-11F; DC-10-10; DC-10-15, DC10-30; DC-10-40; DC-10-10F; DC–10-30F; DC-10-40F in accordance with &quot;Electronic Cables Specialists Master Data List ECS-20347&quot;.</td>
<td>1. ECS-203482 Rev. N/C AFM Supplement for McDonnell Douglas MD-11 Series with Enhanced Mode-S Transponder Upgrade with Flight ID from Transponder Control Panel. 2. ECS-203483 Rev. N/C AFM Supplement for Douglas DC-10 Series Airplane with Mode-S Transponder Upgrade for Elementary Surveillance wir...</td>
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<td>EASA.IM.A.S.01690</td>
<td>TACA INTERNATIONAL AIRLINES TECHNICAL AND TRAINING SERVICES(TTS)</td>
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<td>AIRBUS A320-233</td>
<td>-</td>
<td>A320 Avionics installation.</td>
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<td>04/11/2005</td>
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<td>EASA.IM.A.S.01691</td>
<td>GODERICH AIRCRAFT inc.</td>
<td>-</td>
<td>BEECH B300</td>
<td>-</td>
<td>Artex ELT C406-N Installation.</td>
<td>Avionics Design Services Ltd. Master Drawing List MDL05097 Revision N/C dated 16 September 2005 or later approved revision. Avionics Design Services Ltd. For Continued Airworthiness ICA05097-1 Revision N/C dated 21 September 2005 or later approved revision.</td>
<td>Modification limited to s/n FL-175. Maintenance must be in accordance with above Continued Airworthiness document.</td>
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<td>GODERICH AIRCRAFT inc.</td>
<td>-</td>
<td>BEECH B300</td>
<td>-</td>
<td>Universal UNS 1E Flight Management System Installation.</td>
<td>Avionics Design Services Ltd. Master Drawing List MDL05097 Revision N/C dated 21 September 2005 or later approved revision. Avionics Design Services Ltd. For Continued Airworthiness ICA05097-1 Revision N/C dated 21 September 2005 or later approved revision.</td>
<td>Modification limited to s/n FL-175. Operation must be in accordance with above Continued Airworthiness document.</td>
<td>17/01/2006</td>
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<td>EASA.IM.A.S.01696</td>
<td>AVIONICS SUPPORT GROUP, Inc.</td>
<td>FAA STC ST01191AT</td>
<td>737 ALL MODELS</td>
<td>-</td>
<td>Installation of an Avionics, Inc. Secure Link System for ground only operation.</td>
<td>Avionics Support Group, Inc. Master Data List 25017-MDL. Rev. IR, dated 15th June 2005, or later approved revisions.</td>
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<td>17/01/2007</td>
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<td>EASA.IM.A.S.01698</td>
<td>JETPROP, LLC.</td>
<td>LBA STC SA1169</td>
<td>PIPER PA-46-350P/PYPER PA-46-310P</td>
<td>Installation of the following engines, propellers and associated hardware: Pratt &amp; Whitney PT6A-34 engine limited to 550 SHP at 2200 RPM; PT6A-35 engine limited to 560 SHP at 2190 RPM; PT6A-31 engine limited to 550 SHP at 2200 RPM; Hartzell HC-E4N-31/E8501B-3.5 propeller/haft...</td>
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<td>1. This STC is valid only in connection with FAA STC SA00535AT (EM2 0721/1076) 2. This airplane/engine/propeller combination is limited. The life limit for the modified airplane shall be determined by using the minimum value calculated by the following formula: T = T0 + 0.9 (10145 - T0) Fuselage Life L...</td>
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<td>EASA.IM.A.S.01701</td>
<td>STAR AVIATION, Inc.</td>
<td>REV 1</td>
<td>FAA STC ST02943AT</td>
<td>BOEING 777-300</td>
<td>Installation of Connection by Boeing Next Generation Antenna and Satellite Data System. This Revision reflects the transfer of this STC from Flight Structure, Inc. to Star Aviation, Inc.</td>
<td>- MDL: R44-0242, Revision AM, dated 20th June 2004 for later FAA approved revision. AFM Supplements R44-0397, Revision B, dated 20th June 2008 for the -200 series or R44-0220, revision B, dated 20th June 2006 for the -300 series, or later FAA approved Revision</td>
<td>Amortheiss Limitations Section of the Instruction for Continuous Amortheiss, document numbers R44-0403, revision B, dated 22nd July 2005 for -200 series, R44-0218, revision B, dated 20th October 2004 for -300 series, or later FAA approved revisions. Prior to installation of this modif...</td>
<td>03/05/2007</td>
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<td>EASA.IM.A.S.01703</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01630SE</td>
<td>LEARJET 36</td>
<td>Installation of Honeywell MAHS-800 Multi-Hazard Awareness System</td>
<td>* Learjet Master document List B4/30/150/07 Revision L or later approved revision - configuration</td>
<td>Learjet Master Document List B4/30/150/07 Revision L or later approved revision - configurations 2, 4 and 5 only</td>
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<td>EASA.IM.A.S.01706</td>
<td>CASCADE AVIATION SERVICES INC.</td>
<td>FAA STC</td>
<td>BOEING 737</td>
<td>Installation of Eyebrow Window Deletion</td>
<td>Master Drawing List OSPAC466-MDL, revision A, dated December 8th, 2005, or later FAA approved revision</td>
<td>Compatibility of this design change with other previously approved modifications must be determined by the installer.</td>
<td>14/03/2007</td>
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<td>EASA.IM.A.S.01710</td>
<td>FOKKERSFREIG HTER STC CORP.</td>
<td>FAA STC ST01916LA - EASA IM.A</td>
<td>FOKKER F.27 MARK 050</td>
<td>Installation of Class E Cargo Inteior configuration 24B301-211 in accordance with Boeing Aircraft Corporation Drawing List 24A300, revision B dated November 5, 1996; FAA approved November 7, 1996 or later revision approved in accordance with ED Decision 2004/04/CF dated 10 December 2004. Also conf...</td>
<td>1. Cargo Loading Manual WB-2052 revision 2 or later approved revisions shall be used when loading the aircraft. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibil...</td>
<td>21/05/2008</td>
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<td>EASA.IM.A.S.01711</td>
<td>FOKKERSFREIG HTER STC CORP.</td>
<td>FAA STC ST02600E</td>
<td>FOKKER F.27 MARK 050</td>
<td>Installation of Class E Cargo Inteior configuration 24B301-211 in accordance with Boeing Aircraft Corporation Drawing List 24A300, revision B dated November 5, 1996; FAA approved November 7, 1996 or later revision approved in accordance with ED Decision 2004/04/CF dated 10 December 2004. Also conf...</td>
<td>1. Cargo Loading Manual WB-2052 Rev. C or later approved revisions shall be used when loading the aircraft. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibi...</td>
<td>02/11/2007</td>
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<td>EASA.IM.A.S.01713</td>
<td>ELLIOTT AVIATION</td>
<td>ST02194CH</td>
<td>FALCON 10</td>
<td>Installation of UNS EFI-6691 EFI5</td>
<td>- MDL: 24B363-000 Rev. E, dated 2 January 2007 or later approved revision - AFM supplement (AFMS) as per MDL 24B363-000 Rev. E -- AFMS (AFMS 24B363-000 Rev. A -- &lt; no date &gt; or later approved revision). Off AFMS FMS-2463-000 Rev. B dated 1 December 2006 or later approved revision - Instructions for Continued Amortheiss ICA-2463-00...</td>
<td>Certification basis see CRI A1 IM.A.S.01713</td>
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<td>ELLIOTT AVIATION</td>
<td>FAA STC ST01666CH</td>
<td>FALCON 10</td>
<td>ST01666CH: Installation of a Rockwell Collins TCAS 400Traffic Alert and Collision Avoidance System (TCAS II)</td>
<td>- MDL-2010-576DL Rev. 5 dated 20 October 2006 or later approved revision. AFMS/AFMS (1) for Continued Airworthiness (ICA) as per MDL-2010-576DL Rev. 5.—AFMS PM-261-002 Rev B dated 20 Oct 2006 or later approved revision/or— AFMS FS2010-576 Rev. 1.</td>
<td>Boeing Service Bulletin 747-25-3420 dated June 13 2006 must be installed concurrently with this STC.</td>
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<td>EASA.IM.A.S.01724</td>
<td>JET AVIATION AG, BASEL</td>
<td>-</td>
<td>AIRBUS A320-232</td>
<td>Cabin Interior Outfittting See Master Data List B320/403-001</td>
<td>Flight Manual Supplement B320/403/001-01 Instruction for Continued Airworthiness B320/403-150-01</td>
<td>1. Approval valid only for MSN 2403. 2. The aircraft is certified for private use only, not to be operated for commercial air transport. 3. The aircraft is a No Smoking aircraft. 4. The W-LAN System is limited to Cabin W-LAN, which is not interfaced to systems other than the SATCOM/HSD system...</td>
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<td>EASA.IM.A.S.01731</td>
<td>RAYTHEON AIRCRAFT COMPANY</td>
<td>FAA STC SAB515W</td>
<td>BAE 125-1000A</td>
<td>Installation of Brackets for Cabin Headliner, Window Panels and Side Panels.</td>
<td>In accordance with Arkansas Aerospace Inc Drawing List 1000-016, dated 06/04/92, or later approved revisions.</td>
<td>In accordance with Montana Aerospace Inc Drawing List 1000-020, dated 06/21/92, or later approved revisions.</td>
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<td>RAYTHEON AIRCRAFT COMPANY</td>
<td>FAA STC SAH411SW</td>
<td>BAE 125-1000A</td>
<td>Installation of Electronic Equipment Racks.</td>
<td>In accordance with Arkansas Aerospace Inc Drawing List 1000-020, dated 06/21/92, or later approved revisions.</td>
<td>In accordance with Arkansas Aerospace Inc Drawing List 1000-020, dated 06/21/92, or later approved revisions.</td>
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<td>ARMSTRONG AEROSPACE Inc.</td>
<td>ST02202CH</td>
<td>BOEING 757-200</td>
<td>Installation of Honeywell RESCU 406AF Emergency Locator Transmitter (ELT)</td>
<td>Armstrong Aerospace Master Data List Doc No. DC104-3078-00-Revision B dated November 09 2005 or later approved revision. Armstrong aerospace AFM Supplement Doc No. DC303-3078-00 Revision approved November 04 2005 or later approved revision.</td>
<td>Armstrong Aerospace Master Data List Doc No. DC104-3078-00-Revision B dated November 09 2005 or later approved revision. Armstrong aerospace AFM Supplement Doc No. DC303-3078-00 Revision approved November 04 2005 or later approved revision.</td>
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<td>JET AVIATION AG, BASEL</td>
<td>FAA STC ST00342DE-D</td>
<td>LEARJET 31A</td>
<td>Installation of Single Universal Avionics Corporation UNS-1C FMS</td>
<td>Learjet Inc Master Document List BAS31010008 Rev (=) dated March 5 1998 or later approved revision. AFM Supplement BAS31040105 Revision (=) dated February 16 2000 or later approved revision</td>
<td>Learjet Inc Master Document List BAS31010008 Rev (=) dated March 5 1998 or later approved revision. AFM Supplement BAS31040105 Revision (=) dated February 16 2000 or later approved revision</td>
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<td>PATS AIRCRAFT L.L.C./ PATS AIRCRAFT SYSTEMS/ALOFT AEROARCHITECT S</td>
<td>FAA STC ST00359NY-D</td>
<td>BOEING 757-200</td>
<td>Installation of Honeywell TCAS II system</td>
<td>Master Data List ref. ML-4075160, revision A, dated September 25, 1992 or later approved revision. AFM Supplement ref. TCAS-APP5-0294 revision A or later approved revision</td>
<td>Master Data List ref. ML-4075160, revision A, dated September 25, 1992 or later approved revision. AFM Supplement ref. TCAS-APP5-0294 revision A or later approved revision</td>
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<td>VENTURA AEROSPACE, Inc.</td>
<td>Installation of a 9g Cargo Barrier according to FAA STC ST01806LA in existence EASA STC EASA.IM.A.S.01755 (737-400 added).</td>
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<td>PIEDMONT HIGHTHORNE AVIATION, LLC</td>
<td>Validation of FAA STC ST301CH - Installation of a Honeywell/CAI-674 Traffic Alert and Collision Avoidance System (TCAS I) in accordance with Electro/Onics Top Drawing E0050-100116-01, Rev G, dated January 14, 2004 or later approved revision</td>
<td>Active</td>
<td>03/07/2004</td>
<td>10017534</td>
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<tr>
<td>ROCKWELL COLLINS, Inc.</td>
<td>Installation of Audio Video Entertainment System with Rockwell Collins System No. 1309667-100 and 1301244-001 according FAA STC SA5135NM: Remark: Only 1 EASA STC to be issued by Programmes Department to cover both applications: working hours information from Mr. Anthony M. A.S. 01756, all other...</td>
<td>Active</td>
<td>04/11/2009</td>
<td>10017535</td>
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<tr>
<td>ANORA INTERNATIONAL LLC</td>
<td>Installation of Main Deck Cargo Loading System</td>
<td>Active</td>
<td>27/06/2009</td>
<td>10017536</td>
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<td>BOMBARDIER INC.</td>
<td>Customised Interior Completion Aircraft 20071, in accordance with Aero Consulting Services Ltd. Modification Data Summary ET25000 rev N/C or later approved revision</td>
<td>Active</td>
<td>30/01/2006</td>
<td>10017537</td>
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<tr>
<td>CONCORDE BATTERY CORPORATION</td>
<td>Installation of Cocarde Battery Model No RG-380E/60K sealed lead acid batteries</td>
<td>Active</td>
<td>06/04/2006</td>
<td>10017538</td>
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<td>ANORA INTERNATIONAL LLC</td>
<td>ESA validation of FAA STC ST01644LA revision C to Anora Controlled Drawing List DL83606STC ST01644LA, cargo Loading System, Project Number SA11805LA-T with the operating instructions related to the cargo loading system PIN 83605-51 are described in ANCRA report 8928, This STC is only applicable on Airbus A300-600 aircraft converted to freighter in accordance with EASA A.S. 0030/17/FAA STC ST01431NY the aircraft must be compliant with the instructions for continued airworthiness (see ANCRA reports 8967 revision A and 8284 revision B, or later approved revision)</td>
<td>Active</td>
<td>07/04/2006</td>
<td>10017539</td>
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<td>NORTWEST COMPOSITES</td>
<td>Stowage Bin Extension/Modification to overhaul stowage bin located in the passenger cabin in order to extend the bins for aircraft originally equipped with Heath/Techna Interiors stowage bins.</td>
<td>Active</td>
<td>28/03/2006</td>
<td>10017540</td>
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<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>NORTHWEST COMPOSITES</td>
<td>DCC-943 (MD-83) and DCC-942 (CF6)</td>
<td>Stowage Bin Extension/Modification to overhead stowage bin located in the passenger cabin in order to extend the bins for aircraft originally equipped with C&amp;D interiors stowage bins FAA STC ST00868SE issue on 21st June 2001</td>
<td>- Master Drawing List A074000 rev. J dated 3rd November 2005 or later approved revision. - Instruction for Continued Airworthiness in 20-20-02 rev. 5 dated 3rd November 2005 or later approved revision.</td>
<td>1. Aircraft dated on Master Drawing List A074000 rev. J dated 3rd November 2005 or later approved revision. This approval of this change in type design applies to the above model aircraft only. This approval should not be extended to other aircraft of this model on which other previously approved m...</td>
<td>16/02/2006</td>
<td>Active</td>
<td>10015741</td>
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<td>DUNCAN AVIATION, INC.</td>
<td>CESSNA 525</td>
<td>Installation of Honeywell CAS 67A TCAS II System on Cessna 525. FAA Approval of STC Nr. SA00849WI: 08 September 2000.</td>
<td></td>
<td>The call outs and their priority are to be confirmed as part of each installation.</td>
<td>26/01/2006</td>
<td>Active</td>
<td>10015742</td>
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<tr>
<td>BOMBARDIER AEROSPACE / LEARJET Inc.</td>
<td>LEARJET 31A</td>
<td>Installation of a Honeywell 440, 660 or 880 Radar System</td>
<td></td>
<td>None</td>
<td>20/01/2006</td>
<td>Active</td>
<td>10015743</td>
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<tr>
<td>BEECHCRAFT CORPORATION</td>
<td>HAWKER 800XP</td>
<td>Installation of Collins Integrated Flight Information System (IFIS 5000)and Pro Line 21 Radars</td>
<td>Collins Master Data List AAC-0380-001 Rev., dated July 8, 2004, or later approved revision Collins Airplane Flight Manual Supplement ref. AAC-0386-01 Rev., dated 6th Aug, 2005, or later approved revision Collins Maintenance Manual Supplement ref. AAC-0387-01 Rev., dated May 22, 2005, or later approved revision.</td>
<td>1. Raytheon Aircraft Company Kit 149-3419-0001 is prerequisite prior to installation and approval of STC (EASA.IM.A.S.01768). 2) Raytheon Aircraft Company Kit 149-3420-0001 is prerequisite prior to installation and approval of STC (EASA.IM.A.S.01768). 3) This STC is approved only for the product configurations defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>26/01/2006</td>
<td>Active</td>
<td>10042063</td>
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<td>VISIONSAFE CORPORATION</td>
<td>BOMBARDIER CL-600-2B19 and CL-600-2B16</td>
<td>Installation of Emergency Vision Assurance System (EVAS TM)</td>
<td>1) Midcoast MDL 34-4169G001, Rev. B, or Vision Safe Corporation MDL 176, Rev. C or later FAA approved revisions 2) AFMS to Canadair CL-600 AFM STC No. ST01012WI was prerequisite prior to installation and approval of STC (EASA.IM.A.S.01768).</td>
<td>1) Certification Basis: applicable requirements of r. TCDS82 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>22/12/2005</td>
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<td>HONEYWELL INTERNATIONAL INC.</td>
<td>BOEING MD-90-30 and BOEING DC-9-87 (MD-87) and BOEING DC-9-83 (MD-83) and BOEING DC-9-81 (MD-81)</td>
<td>FAA STC ST00491SE &quot;Installation of the Enhanced Ground Proximity Warning System&quot;</td>
<td>Master Drawing List 960-2227-000 Rev. AY dated 7 June 2005AFM as indicated in Master Drawing List 950-2227-007 Rev. AY dated 7 June 2005.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>05/03/2007</td>
<td>Active</td>
<td>10015745</td>
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<td>AEROCON ENGINEERING COMPANY</td>
<td>JETSREAM 4100</td>
<td>Installation of Smoke Detection &amp; Fire Suppression Systems for Conversion of Class D Cargo Compartments to Class C</td>
<td></td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>17/02/2006</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<tr>
<td>EASA.IM.A.S.01781</td>
<td>NATIONAL AIRCRAFT SERVICES, INC.</td>
<td>FAA STC ST0183CH</td>
<td>F4-605R F4-622R C4-605 VAR F B4-603 B4-620 B4-605R B4-622R AIRBUS A310-304 AIRBUS A310-222 AIRBUS A300 MODEL B4-2C AIRBUS A300 MODEL B4-103 AIRBUS A300 MODEL B2K-3C AIRBUS A300 MODEL B2-203 AIRBUS A300 MODEL B2-1C AIRBUS A300 MODEL B2-1A</td>
<td>EASA validation of F4 STC ST0183CH Reinforced Flightdeck Door and Venting System Installation of reinforced flightdeck door and venting system defined in Drawing List FDS-DL16, Revision M, dated August 26, 2003, or later FAA approved revisions, in accordance with Cockpit Door System Installation Ma...</td>
<td>The National Aircraft Service, Inc Airplane Flight Manual Supplement Number FDS-19-AFM, FAA approved on September 15, 2003, or later EASA Approved Revisions, is required with this STC.</td>
<td>10/04/2006</td>
<td>Active</td>
<td>10115742</td>
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<td>1. This flightdeck door modification has not been shown to provide smoke penetration and evacuation protection as required by 14 CFR 25.831. An EASA approved smoke barrier/curtain, installed behind the flightdeck bulkhead and in front of the cargo compartment, is required as part of the airplane type...</td>
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<td>EASA.IM.A.S.01782</td>
<td>GULFSTREAM AEROSPACE Corp.</td>
<td>REV 2</td>
<td>FAA STC ST103015C-D</td>
<td>GULFSTREAM 200</td>
<td>Installation of Executive Cabin Interior in accordance with Index List GA22048001, Rev A, dated 3-23-2005 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/44/CF (or subsequent revisions of this decision).</td>
<td>Gulfstream Aerospace Index List GA22048001, or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/44/CF (or subsequent revisions of this decision).Approved Flight Manual Supplement GA212234M105, Rev. None, dated 3/23/2005 or later revisions of the...</td>
<td>Compatibility of this design change with previously approved modifications must be determined by the installer.</td>
<td>03/03/2008</td>
<td>Active</td>
<td>10115749</td>
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<tr>
<td>EASA.IM.A.S.01785</td>
<td>JET AVIATION AG, BASEL</td>
<td>-</td>
<td>BOEING 737-800</td>
<td>Installation of modified customized seats</td>
<td>Flight Manual Supplement Doc No BBJ33473-201-06</td>
<td>Flight Manual Supplement Doc No BBJ33473-201-06</td>
<td>1. Modification limited to Aircraft S/N 334732. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>30/01/2006</td>
<td>Active</td>
<td>10115752</td>
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<tr>
<td>EASA.IM.A.S.01786</td>
<td>HAECO AIRFRAME SERVICES, LLC.</td>
<td>FAA STC ST03200AT</td>
<td>BOEING 737-800</td>
<td>Complete interior reconfiguration in accordance with TIMCO Aviation Services, Inc MDL report 04T101L500, revision &quot;D&quot;, dated December 08, 2005, or later EASA approved revision</td>
<td>- MDL report 04T101L500, revision &quot;D&quot;, dated December 08, 2005. - TIMCO report 04T110R014, Instructions for Continued Airworthiness - Interior reconfiguration, Revision &quot;RT&quot;, dated March 01, 2005, or later EASA approved revisions.</td>
<td>- MDL report 04T101L500, revision &quot;D&quot;, dated December 08, 2005. - TIMCO report 04T110R014, Instructions for Continued Airworthiness - Interior reconfiguration, Revision &quot;RT&quot;, dated March 01, 2005, or later EASA approved revisions.</td>
<td>20/07/2006</td>
<td>Active</td>
<td>10115753</td>
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### List of EASA Supplemental Type Certificates

#### EASA STC No. 01758
- **STC Holder:** FIEL AVIATION EAST, LTD.
- **Certification basis:** Master Document List
- **Models:** BOMBARDIER DHC-8-315
- **Date of issue:** 04/06/2006
- **Status:** Active

**Description:**

Installation of IARUS Sky Connect Tracker SOP and Satellite Communication System

**Limitations/Conditions:**

- This approved is only applicable to the type/model of aeronautical product specified therein. Prior to incorporating this modification, the installer shall establish that the interrelationship between this change and any other modification(s) incorporated will not adversely affect the airworthiness...

#### EASA STC No. 01790
- **STC Holder:** HONEYWELL INTERNATIONAL INC.
- **Certification basis:** Master Document List
- **Models:** BOEING 747-400
- **Date of issue:** 03/03/2006
- **Status:** Active

**Introduction of Honeywell International Inc. Main/Nose Wheel pn. 2615901-1 and Carbon Brake pn. 2615902-1 in accordance with Honeywell Service Bulletin 2615801-33-001_2615802-33-001 Revision New or later FAA-approved revision as replacement for Goodrich Brake pn.2-1515-1 and/or Wheel pn. ...

#### EASA STC No. 01791
- **STC Holder:** ELLIOTT AVIATION
- **Certification basis:** Master Document List
- **Models:** BEECH 300C
- **Date of issue:** 03/06/2006
- **Status:** Active

**Description:**

Installation of Honeywell Digital Flight Data Recorder System (DFDR)

**Limitations/Conditions:**


- Limited to the aircraft types listed above with Proline 21 Avionics Suite installed.

#### EASA STC No. 01793
- **STC Holder:** ASSOCIATED AIR CENTER
- **Certification basis:** Master Document List
- **Models:** AIRBUS A319-115
- **Date of issue:** 13/06/2006
- **Status:** Active

**Description:**

VIP Cabin modification

**Limitations/Conditions:**

- Master Document List (MDL), ref. 23500000, revision A, dated June 8, 2006 or later EASA approved revision.FAFA Approved Aircraft Flight Manual Supplement (AFM-6), ref. 2350021060, revision R, dated June 9, 2006 or later EASA approved revision.Instructions for Continued Airworthiness (ICAWS), Chapter 3

- This STC is limited to the A319 CJ MSN 2952 Instructions for Continued Airworthiness (ICAWS), Chapter 3 "Airworthiness Limitations" ref. 2350050090, revision A dated June 8, 2006 or later EASA approved revision.

#### EASA STC No. 01795
- **STC Holder:** LAKE CENTRAL AIR SERVICES
- **Certification basis:** TCCA STC
- **Models:** BOEING 737-200, 200C, -300, -400, -500
- **Date of issue:** 18/04/2006
- **Status:** Active

**Description:**

Horizontal Stabilizer Attachment Fitting PN LC-2200-21 TCCA STC SA-05-71 Issued on 8th July 2005

**Limitations/Conditions:**

- Parts must be manufactured in accordance with Lake Central Air Services Drawing LC-2200-21, Rev. 0, dated July 06, 2005, and Lake Central Air Services Manufacturing Procedures MP-LC-2200-21, Rev. B, dated July 06, 2005, or later approved revisions.- Installation must be performed in accordance with...

#### EASA STC No. 01796
- **STC Holder:** NASCO BRAKE, INC.
- **Certification basis:** Master Document List
- **Models:** BOEING 737-700, 800, -900
- **Date of issue:** 31/05/2006
- **Status:** Active

**Description:**

Installation of brake components listed on Nasco Aircraft Brake, Inc drawing 737101, Revision , dated August 11, 2005 for use in Goodrich brake assemblies 2-1474-3, -4, -5 and -7 in accordance with Nasco Configuration Control Drawing NCC007, Revision B, dated November 7, 2005 or later approved revision...
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<th>FAA STC No/STC ref</th>
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<tr>
<td>EASA.IM.A.S.01791</td>
<td>ROCKWELL COLLINS Inc.</td>
<td>FFA STD ST01903LA</td>
<td>BOEING 777-200</td>
<td>Total Entertainment System Index List 1304641 rev 9 dated 02 August 2006. Modification comprises a software upgrade to the VMOD, re-identification of the VMOD and re-installation in the Video Control Centre.</td>
<td>FAA STC ST01903LA dated 02 August 2006</td>
<td>FAA STC ST01903LA dated 02 August 2006 Rev 0 (\text{VMOD upgrade, ref Rockwell Collins System List 1310328-001 rev (0) Continued Airworthiness Document, 130944 rev AF.}</td>
<td>Limited to 977-200 models operated by British Airways plc.</td>
<td>19/12/2007</td>
<td>Active</td>
<td>10101751</td>
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<tr>
<td>EASA.IM.A.S.01798</td>
<td>WIGGINS AIRWAYS, INC.</td>
<td>FFA STC ST005280</td>
<td>RAYTHEON 1900D</td>
<td>Validation of FAA STC ST005280 Incorporation of a RockwellCollins TCAS 94 System (TCAS II) with TTR-951 Change 7.0 R/T, dual TVS-900 TARAV/VI Indicators dual TDR-94D Mode S Transponders, CTL-92T TCAS Control and CTL-92 Transponder Control.</td>
<td>FAA approved airworthiness dates</td>
<td>FAA approved airworthiness dates</td>
<td>This STC does not comply with requirements for Mode S elementary surveillance or for Mode S enhanced surveillance.</td>
<td>05/04/2006</td>
<td>Active</td>
<td>10101752</td>
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<td>EASA.IM.A.S.01806</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC SAB2705W-D</td>
<td>FALCON 900EX</td>
<td>Installation of the Miscellaneous Structures (General / Mechanical)</td>
<td>Drawing List Report No. F9EX-01008 Rev. A dated 09-09-97 or later EASA approved Revisions.</td>
<td>Drawing List Report No. F9EX-01008 Rev. A dated 09-09-97 or later EASA approved Revisions.</td>
<td>08/02/2006</td>
<td>Active</td>
<td>10101755</td>
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<tr>
<td>EASA.IM.A.S.01809</td>
<td>SIMMONDS PRECISION PRODUCTS, Inc</td>
<td>FAA STC ST000480</td>
<td>BOEING 737-500 BOEING 737-400 BOEING 737-300 BOEING 737-200 BOEING 737-100</td>
<td>Installation of a Simmonds Precision 90203 series Fuel Quantity Harness in accordance with Simmonds Precision Installation Document No E-3203, dated May 24, 1995, or later approved revision</td>
<td>FAA approved airworthiness dates</td>
<td>FAA approved airworthiness dates</td>
<td>Compatibility of this design change with previously approved modifications shall be determined by the installer.</td>
<td>22/01/2007</td>
<td>Active</td>
<td>10101758</td>
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<td>EASA.IM.A.S.01810</td>
<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>ST0304AT</td>
<td>BAE JETSTREAM 4101</td>
<td>Installation of Honeywell MX VI Enhanced Ground Proximity Warning System (EGPWS)</td>
<td>FAA approved airworthiness dates</td>
<td>FAA approved airworthiness dates</td>
<td>Compatibility of this design change with previously approved modifications shall be determined by the installer.</td>
<td>31/05/2006</td>
<td>Active</td>
<td>10101759</td>
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## List of EASA Supplemental Type Certificates

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<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
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<th>Limitations/Conditions</th>
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<td>EASA.IM.A.S.01817</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03196AT-D</td>
<td>GULFSTREAM GV-S/P</td>
<td>Completion and installation of the custom interior on Aircraft Serial 5004</td>
<td>In accordance with Gulfstream Aerospace Corporation Master Index List GC314915004, Revision E, dated March 31, 2006, after the FAA approved Airworthiness Directives.</td>
<td>FAA Approved Flight Manual Supplement, GC314915004, Revision NC, dated March 31, 2006, or later approved revisions.</td>
<td>Aircraft Serial 5004 Only</td>
<td>21/04/2006</td>
<td>Active</td>
<td>10015773</td>
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<td>EASA.IM.A.S.01818</td>
<td>JET AVIATION AG, BASEL</td>
<td>-</td>
<td>AIRBUS A320-233</td>
<td>CAR TM System Installation</td>
<td>Instructions for Continued Airworthiness BA32021701-150-01</td>
<td>FAA Approved Flight Manual Supplement, GC31501M014, Revision NC, dated March 31, 2006, or later approved revisions.</td>
<td>The aircraft maximum number of passengers is reduced by 7 (seven) passengers.</td>
<td>30/06/2006</td>
<td>Active</td>
<td>10015774</td>
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<tr>
<td>EASA.IM.A.S.01825</td>
<td>JET AVIATION AG, BASEL</td>
<td>-</td>
<td>GULFSTREAM 1159 (G-II)</td>
<td>Weather Radar Installation</td>
<td>Declaration of Compliance Project B784008Master Data List BTA4005-001</td>
<td>FAA Approved Airplane Flight Manual Supplement for Gulfstream 1159 (G-II) 05/06/2006</td>
<td>05/06/2006</td>
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<td>EASA.IMA.S.01826</td>
<td>A.A.E., LTD.</td>
<td>REV 1</td>
<td>FAA STO114W</td>
<td>737-300, 737-400, 737-500</td>
<td>Installation of Smoke Detection and Fire Suppression System</td>
<td>FAA/LM Master Drawing List TFA10-0061, Revision H, dated 01 December 2005 or later approved revision.AFM Supplement ref. TF90-0063, TF90-0064, TF90-0065, TF90-0066 dated 20 November 2002. TF90-0067, TF90-0088 dated 20 June 2005, TF90-0069, TF90-0089, TF90-0090 dated 16 December 2005 or L.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>26/11/2002</td>
<td>Active</td>
<td>10/17/1761</td>
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<td>EASA.IMA.S.01842</td>
<td>S-TEC CORPORATION</td>
<td>REV 1</td>
<td>FAA STC SA09500AC-D</td>
<td>VULCANAIR P68 C VULCANAIR P68 B</td>
<td>Installation of S-TEC System 55 two axis automatic flight guidance system, model ST-665.</td>
<td>FAA/DAS approved PHV Automatic Flight Guidance System for Partenavia P68B, P68C and PMBC-TC with S-TEC System 55 two axis automatic flight guidance system P/N 891390, Revision 1 or later FAA/DAS approved revisions.</td>
<td>1. Also eligible on the above data (28V System).</td>
<td>10/06/2008</td>
<td>Active</td>
<td>10/17/1766</td>
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<td>EASA STC No.</td>
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<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>EASA.IM.A.S.01843</td>
<td>MASS SYSTEMS A UNIT OF AMERON GLOBAL, INC.</td>
<td>FAA STC ST01708LA</td>
<td>SD3-60 SHERPA SD3-30 SD3-60 SD3-SHERPA</td>
<td>Installation of MASS Systems’ fire extinguisher, P/N M57331-009, as a replacement for the Graviner fire extinguisher, P/N 57331-009 in the above-noted airplane model. This fire extinguisher is in accordance with MASS Systems’ Drawing No. M57331-009, “Fire Extinguisher, 52 cu-in”, Revision B, dated No...</td>
<td>see above</td>
<td>The approval of this change in type design applies to the above-noted airplane model. This approval should not be extended to other specific airplanes of these model series on which other than noted previously approved modifications are incorporated, unless it is determined that the interrelationship...</td>
<td>16/04/2007</td>
<td>Active</td>
<td>10015757</td>
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<td>EASA.IM.A.S.01844</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST00508DE-D</td>
<td>BOMBARDIER CL-660-2B16</td>
<td>Installation of Collins PCD-3000 Data Loader Interface System</td>
<td>Learjet Inc Master Document List BAS65010085 Revision (-) dated January 22, 2002 or later approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>20/01/2006</td>
<td>Active</td>
<td>10015788</td>
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<td>EASA.IM.A.S.01845</td>
<td>ARMSTRONG AEROSPACE Inc.</td>
<td>ST01797CH</td>
<td>AIRBUS A321 AIRBUS A320</td>
<td>Installation of a Honeywell RESCU 406 AF Emergency Locator Transmitter (ELT)</td>
<td>Master Data List (MDL), ref. DC104-1020-00, revision F, dated June 17, 2005 or later EASA approved revision.</td>
<td>Instructions for Continued Airworthiness (ICA), ref. DC113-1020-00, Section 2, revision D, dated June 17, 2005 or later EASA approved revision.</td>
<td>17/06/2006</td>
<td>Active</td>
<td>10015789</td>
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<td>EASA.IM.A.S.01848</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01595M-D</td>
<td>BOMBARDIER CL-660-2B16</td>
<td>Installation of a Collins VHF-22C Comm System with CTL-22C.</td>
<td>FAA STC No. SA8139NM-D - Learjet Inc. MDL No. B5AS65010085 Revision J, dated 9 March 2000 or later approved revision. Learjet Inc. AFM Supplement No. B5AS6501111, Revision A, dated 3 July 1999 or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>07/05/2007</td>
<td>Active</td>
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<td>EASA.IM.A.S.01849</td>
<td>DECA AVIATION ENGINEERING Ltd.</td>
<td>TCCA LSTC O-LS02-653/0 ISSN 3</td>
<td>A318-112, A320-214</td>
<td>Passenger seat leather dress cover replacement</td>
<td>DECA Modification summary n° MS20313, revision 3, dated December 22, 2003 or later EASA approved revision.</td>
<td>This STC is limited to: A319 MSN 1846, A319 MSN 1598 and A320 MSN 1698</td>
<td>12/05/2006</td>
<td>Active</td>
<td>10015791</td>
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<td>EASA.IM.A.S.01851</td>
<td>O&amp;B AIRCRAFT MODIFICATIONS, INC.</td>
<td>FAA STC SA165NE</td>
<td>CESSNA 172P CESSNA 172N CESSNA 172M CESSNA 172I CESSNA 172H CESSNA 172Q CESSNA 172F</td>
<td>Installation of an 18 gallon fuel tank in the aft baggage compartment, and fuel transfer system.</td>
<td>FAA STC No. SA165NE - Installation Drawing List 172100, Report No O&amp;B-1012, Revision A, dated June 1 1989, or later approved revision. Eligible on aircraft serial numbers as identified on the Approved Flight Manual Supplements listed below.-- O &amp; N Supplemental Fig...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>29/08/2006</td>
<td>Active</td>
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<td>Limitations/Conditions</td>
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<td>Associated Technical Documentation</td>
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<td>EASA:IM.A.S.01853</td>
<td>MASS SYSTEMS, A UNIT OF AMERON GLOBAL, INC.</td>
<td>00961LA</td>
<td>SIKORSKY S-76B</td>
<td>CESSNA 550</td>
<td>Installation of MASS Systems Fire Extinguisher Cartridge as replacement for Walter Kidde Aerospace fire extinguisher cartridges in fire extinguisher bottles installed in the airplane models noted on the EASA approved model list dtd 02 April 2008 (which excludes aircraft models not validated by EASA). These fire extinguisher cartridges are in accordance with the MAAS Systems Hermetic Cartridges Kidde Master Drawing List No. 30980-1 “Slug Thrower Cartridges,” Rev N/C, dtd 28 June 2000 or later FAA and EASA approved revisions.</td>
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<td>08/04/2008</td>
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<td>EASA STC no.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.01854</td>
<td>MASS SYSTEMS, A UNIT OF AMERON GLOBAL, INC.</td>
<td>FAA STC ST08886LA</td>
<td>DATED 24/01/2007 AS PER APPROVED MODEL LIST</td>
<td>Installation of Mass Systems, Inc. fire extinguisher cartridges as replacements for Pacific Scientific fire extinguisher cartridges in fire extinguisher bottles installed in the airplane models noted on the EASA Approved Model List dated 24th January.</td>
<td>FAA STC EASA.IM.A.S.01854</td>
<td>STC SA09007AC-D</td>
<td>STC SA01962CH</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description” and “Associated Technical Documentation”.</td>
<td>24/01/2007</td>
<td>Active</td>
<td>10115713</td>
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<td>EASA.IM.A.S.01855</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST07171LA-D</td>
<td>BOMBERGER BD-100-1A10 (CHALLENGER)</td>
<td>Installation of a lavatory Seat Modification - Belt replacement, in accordance with Master Drawing List BAS68810002 revC</td>
<td>CAA-NL report No. LT/B TCP06.531118</td>
<td>SATCOM System</td>
<td>STC SA01962CH</td>
<td>Certification basis: a deviation from 25.789(b) (has been granted for personal injury criteria testing of side single occupancy seats (Learjet CFI D-3)</td>
<td>07/04/2006</td>
<td>Active</td>
<td>10115766</td>
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<tr>
<td>EASA.IM.A.S.01858</td>
<td>OFEK HOLDING CORPORATION</td>
<td>FAA STC SAI1415CH</td>
<td>PIPER PA-31T</td>
<td>Installation of camera provisions in fuselage floor cut-out</td>
<td>FAA STC ST1193LA-T</td>
<td>SATCOM High-Gain Antenna</td>
<td>STC SA01962CH</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>30/01/2006</td>
<td>Active</td>
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<td>EASA.IM.A.S.01859</td>
<td>OFEK HOLDING CORPORATION</td>
<td>FAA STC SAI1961CH</td>
<td>PIPER PA-31T</td>
<td>Installation of OFEK Holding Company camera provisions installed pursuant to STC SAI1415CH</td>
<td>FAA STC ST1193LA-T</td>
<td>SATCOM High-Gain Antenna</td>
<td>STC SA01962CH</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>30/01/2006</td>
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<td>10115768</td>
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<td>EASA.IM.A.S.01863</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC ST00200LA</td>
<td>BOEING 767-300 SERIES</td>
<td>Installation of Honeywell MCS-6000 SATCOM System</td>
<td>FAA STC ST1193LA-T</td>
<td>SATCOM High-Gain Antenna</td>
<td>STC SA01962CH</td>
<td>This installation must be installed in conjunction with Honeywell STC EASA.IM.A.S. 02276 for the installation of the SATCOM High-Gain Antenna and Diplexer/Low Noise Amplifier (DNAP).</td>
<td>08/03/2007</td>
<td>Active</td>
<td>10115799</td>
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<tr>
<td>EASA.IM.A.S.01864</td>
<td>DELTA AIR LINES, INC</td>
<td>SAI30550</td>
<td>BOEING 767-200</td>
<td>Installation of a Delta modified forward bathroom bearing housing &amp; Delta manufactured bearing retainer bolt to allow for lubrication of the respective joints.</td>
<td>FAA STC ST1193LA-T</td>
<td>SATCOM High-Gain Antenna</td>
<td>STC SA01962CH</td>
<td>Delta E.O. 1.65441-3 dated August 08 1992 or later approved revision.</td>
<td>01/06/2006</td>
<td>Active</td>
<td>10115800</td>
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<td>EASA.IM.A.S.01866</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SAI95927AC-D</td>
<td>TWIN COMMANDER AIRCRAFT 600B, TWIN COMMANDER AIRCRAFT 690A, TWIN COMMANDER AIRCRAFT 690</td>
<td>Installation of S-TEC Digital Flight Control System (autopilot) Part Numbers 01294 or 01298-01-01-002 or 01-304-04-01-008, (MAOC 2100) (ST-802) in Twin Commander 690, 690A and 690B, also if modified by any of the following STC’s: STC SA3861NM (Garrett TPE 331-10T-515 engines), STC SA3861NM (&quot;Q&quot; tip propellers), or STC SAI133962 (winglets), installed in conjunction with Master Drawing No. 01544-06-01-002, Rev C, or later approved revision.</td>
<td>Master Data List No. 921219, Rev. B or later FAA approved revision. Installation Instructions ST-902-1, Rev. C, or later released revision.</td>
<td>SATCOM System</td>
<td>STC SA01962CH</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>20/10/2006</td>
<td>Active</td>
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<td>EASA.IM.A.S.01867</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SAI9807AC-D</td>
<td>CESSNA 182Q, CESSNA 182P</td>
<td>Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System Model ST-687.</td>
<td>FAA STC SA3861NM (&quot;Q&quot; tip propellers), or STC SAI133962 (winglets), installed in conjunction with Master Data List No. 921219, Rev. B or later FAA approved revision. Installation Instructions ST-902-1, Rev. C, or later released revision.</td>
<td>SATCOM System</td>
<td>STC SA01962CH</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>10/06/2008</td>
<td>Active</td>
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<td>EASA STC no.</td>
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<td>Foreign/NA A STC ref</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.01868</td>
<td>AIRCRAFT SAFETY EQUIPMENT</td>
<td>5A278E5</td>
<td>SA000048C-D</td>
<td>MD-11, MD-11F</td>
<td>Smoke Barrier Cargo Compartment McDonnell-Douglas (Boeing) MD-11F, aircraft serial no. 44145, Installation of a modified barrier in the Class &quot;E&quot; cargo compartment in accordance with FAA STC SA0078E5, reissued on September 24, 2004 1. Aircraft Safety Equipment, Memphis, TN. Compliance Checklist: Smoke Curtain MD-11 (no date); 2. Aircraft Safety Equipment, Memphis, TN. Cargo Compartment Smoke Barrier Smoke Penetration Test Procedure 91250901, dated 31-01-1992.3. Ship Shape Custom Products, Mooresville, NC. Flammability Test Repor...</td>
<td>1. The Aircraft Flight Manual should be amended in order to make sure that before each flight the integrity of the smoke barrier is checked; 1) the smoke barrier should be without any tears; 2) the OEM crown area smoke barrier should be without any tears; 3) all gaps around the smoke barrier shou...</td>
<td>14/11/2006</td>
<td>Active</td>
<td>10105302</td>
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<tr>
<td>EASA.IM.A.S.01870</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>5A000048C-D</td>
<td>ST007055E</td>
<td>BOEING 747-SP, BOEING 747-300, BOEING 747-10F, BOEING 747-100</td>
<td>Installation of Honeywell Enhanced Ground Proximity Warning System (EGPWS).</td>
<td>Master drawing 060-4198/054, revision BF, dated 13-11-2008 or later revisions of the above listed documents approved by EASA in accordance with EASA Decision 2004/454/EC (or subsequent revisions of this decision).</td>
<td></td>
<td>06/09/2007</td>
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<td>10105305</td>
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<td>EASA.IM.A.S.01874</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>5A000048C-D</td>
<td>ST00440SE</td>
<td>A310-300, A310-300, A300B4-600, A300B4-600, A300B2K-3C, A300B4-2C</td>
<td>Installation of Honeywell EGPWS on Airbus A300/A310/A300-600 aircraft. This STC installs a Honeywell Enhanced Ground Proximity Warning System (EGPWS).</td>
<td>-Master Drawing List, ref: 960-2227-005, Revision A/E, dated 11 October 2005 (or any later EASA approved revision). -Aircraft Flight Manual Supplement listed for the installation. -Instruction for Continued Airworthiness listed for the installation.</td>
<td>1. Limitations on the STC EASA.IM.A.S.01874 are as the ones defined through the original FAA STC ST0440SE, at its latest amendment dated 27 May 2004 (or any later EASA approved revision). 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the pa...</td>
<td>24/05/2007</td>
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<td>EASA.IM.A.S.01876</td>
<td>GULFSTREAM AVIATION CORPORATION</td>
<td>5A000048C-D</td>
<td>ST00552CH-D</td>
<td>GIV-X</td>
<td>Installation of Class B Baggage Compartment.</td>
<td>In accordance with Gulfstream Aerospace Corporation Master Drawing ListGCA25040655, Revision B, dated June 10, 2005, or later approved revisions.Gulfstr...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/equipment configurations shall be determined by the installer.</td>
<td>17/03/2006</td>
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<td>10105308</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.01877</td>
<td>WORLDWIDE AIRCRAFT SERVICES Inc./ D/B/A WORLDWIDE AIRCRAFT SERVICES</td>
<td>FAA STC ST00786W</td>
<td>FAIRCILD</td>
<td>FAIRCILD</td>
<td>FAIRCILD</td>
<td>FAIRCILD</td>
<td>Installation of Fairchild Digital Flight Data Recorder F800 or F1000 in Fairchild aircraft SA227-AT, -AC, BC, CC, DC.</td>
<td>Worldwide Aircraft Services Certification Drawing List No. DL227-31-06 Revision A, dated May 14, 1999 or later approved revision.</td>
<td>The FAA approved Aircraft Flight Manual Supplement must be onboard the aircraft at all times. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>02/11/2007</td>
<td>Active</td>
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<td>EASA.IM.A.S.01881</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA STC SA86-2</td>
<td>BD-700-1A10</td>
<td>Main Entry Door Handrail Modification/Installation of TCCA STCSA86-2MDL. BAS87610036 REV NC or later TCCA approved revisions. TCCA SU-C: C-06-1053 SAD8-2 dated January 23, 2006</td>
<td>IFCA: MMS382701R or later TCCA approved revisions</td>
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<td>23/01/2007</td>
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<td>Certification basis</td>
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<td>EASA.IM.A.S.01884</td>
<td>LANDMARK AVIATION</td>
<td>SA0086CH</td>
<td>CESSNA 525</td>
<td>Installation of an Allied Signal two tube EFIS-50 on Co-pilot’s panel with Honeywell F3600 Flight Director in accordance with ElectroSonics Drawing No. ES525-720250-01, Revision 1, dated February 2, 1999 or later FAA approved revision.</td>
<td>see above</td>
<td>1. Compatibility of this design change with previously approved modifications must be determined by the installer.2. The aircraft must be maintained in accordance with FAA Approved Maintenance Manual Supplement, document number ES525-720250-200. Initial Release, dated February 02, 1999, or later FAA...</td>
<td>05/10/2006</td>
<td>Active</td>
<td>10015814</td>
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<tr>
<td>EASA.IM.A.S.01887</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA STC C-LSA06-178D</td>
<td>BOMBARDIER CL-600-2B16 (CL-604</td>
<td>Aircraft Completion</td>
<td>Aero Consulting Services Modification Data Summary ES21000 Revision N/C or later approved revision</td>
<td>Aero Consulting Services AFM Supplement E021090 Revision N/C or later approved revision</td>
<td>--</td>
<td>27/06/2006</td>
<td>Active</td>
<td>10015816</td>
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<td>EASA.IM.A.S.01888</td>
<td>LEARJET Inc.</td>
<td>REV. 1</td>
<td>FAA ST00203BO</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>EGPWS Software Upgrade Installation of Cockpit Speaker Mute</td>
<td>FAA SOC dated 06 May 2008 and Bombardier Aerospace BusinessAviation Services Master Document List, document number BASS70100, Revision F, dated May 04, 2007 or later approved revision.</td>
<td>03/06/2008</td>
<td>Active</td>
<td>10015817</td>
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<tr>
<td>EASA.IM.A.S.01889</td>
<td>AIRWORKS</td>
<td>FAA STC ST01576LA</td>
<td>146AVRO 146 SERIES</td>
<td>Installation of Airwork Cockpit Door Surveillance Systems</td>
<td>FAA STC ST01576LA, AirWorks Index List Document No. IL23003 Revision A dated 15 2004 or later approved revision</td>
<td>AirWorks/AFM Supplement Document No. AMI509050 Revision Original dated January 6 2004 or later approved revision.</td>
<td>04/01/2007</td>
<td>Active</td>
<td>10015818</td>
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<tr>
<td>EASA.IM.A.S.01891</td>
<td>AMSAFE, Inc.</td>
<td>FAA STC ST01866LA</td>
<td>JETSTREAM 4100 SERIES</td>
<td>Installation of AMSAFE Aviation Inflatable Restraint (AIR) System on Front Row Passenger Seats of the Jetstream 4100 Series</td>
<td>AMSAFE Master Data List ES09060 Revision NC dated January 42006 or later approved revision.</td>
<td>AMSAFE Master Data List ES09060 Revision NC dated January 42006 or later approved revision.</td>
<td>Not valid for Jetstream Model 4124</td>
<td>17/05/2006</td>
<td>Active</td>
<td>10015819</td>
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<tr>
<td>EASA.IM.A.S.01892</td>
<td>AEROMECH Inc.</td>
<td>ST01413SE</td>
<td>HAWKER HS 125-700B</td>
<td>HAWKER HS 125-700A</td>
<td>Approval of Reduced Vertical Separation Minimum (RVSM).</td>
<td>In accordance with Aeromech Document No AIM-QD-H700-010 Rev C, dated December 7, 2005, or later approved revisions. Operation to be in accordance with Aeromech Airplane Flight Manual Supplement ref AIM-AFMS-H700-0417 Rev A, dated, May 18, 2005, or later approved revisions.</td>
<td>1. Only applicable to Aircraft Serial 257037.2. This STC must be installed after Avionics Mobile STC ref AM1518. The installation is to be completed by the embodiment of Avionics/Mobile Minor change AM1532 installation of RVSM AOM’s.</td>
<td>17/10/2006</td>
<td>Active</td>
<td>10015820</td>
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<tr>
<td>EASA.IM.A.S.01893</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST01518LA-O</td>
<td>GULFSTREAM GV-SP</td>
<td>Completion and installation of the custom interior on Aircraft Serial No 5040</td>
<td>In accordance with Gulfstream Aerospace Corporation Master Index List GC52008421. Revision E, dated February 11, 2005, or later approved revisions FAA approved Flight Manual Supplement, GC51 658405 Revision NC, dated October 26, 2004, or later approved revisions.</td>
<td>Aircraft Serial 5040 Only</td>
<td>24/08/2006</td>
<td>Active</td>
<td>10015821</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>EASA.IM.A.S.01896</td>
<td>KELOWNA FLIGHTCRAFT Ltd</td>
<td>SA04-32</td>
<td>BOMBARDIER CL-215-6B11</td>
<td>TCCA STC SA04-32 issue 35PF-2500 Honeywell AFC3 and Universal U1K 1% FMS</td>
<td>- MDL K-16-0000-02- AFMS K126-102-014 A, K126-102-014 A and K126-102-014,3A</td>
<td>None</td>
<td>17/07/2006</td>
<td>Active</td>
<td>1016824</td>
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<tr>
<td>EASA.IM.A.S.01897</td>
<td>NORTHROP GRUMMAN GUIDANCE AND ELECTRONIC SYSTEMS, INC.</td>
<td>REV 1</td>
<td>FAA STC SA027NM</td>
<td>B747-287</td>
<td>Replacement of old inertial navigation equipment with newer technology</td>
<td>1. Top: Installation Drawing LTN-92 INS in B747-287B, Litton Document 454005-0. Rev. NIC.2. Engineering Authorization. Litton Document 464428, Rev. NIC. C3. Flight Manual Supplement to Boeing B747-200 Series Airplane, Litton Document 464427, Rev. NIC. C4. FAA Supplemental Type Certificate SA...</td>
<td>Litton Airplane Flight Manual Supplement, Doc No 464427, NIC. FAA approved on August 8, 1994, or later approved revisions, is required as part of this modification. This STC is approved only for the Boeing 747-287B model airplanes, and it applies only when three (3) Litton LTN-92 INSs have been used to...</td>
<td>28/11/2006</td>
<td>Active</td>
<td>1016825</td>
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<td>EASA.IM.A.S.01904</td>
<td>DUNCAN AVIATION, INC.</td>
<td>ST0932WI</td>
<td>LEARJET 36A LEARJET 36 LEARJET 35A LEARJET 35 LEARJET 31</td>
<td>Installation of Universal Avionics TAWS</td>
<td>None</td>
<td>None</td>
<td>11/06/2006</td>
<td>Active</td>
<td>1016826</td>
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<td>EASA.IM.A.S.01919</td>
<td>DECA AVIATION ENGINEERING LIMITED</td>
<td>REV 2</td>
<td>TCCA LSTC O-LA04-656-D ISS1</td>
<td>DHC-6-300</td>
<td>Installation of an ASIRAS Radar System in a DHC-6-300 aircraft in accordance with Modification Summary for ASIRAS Radar System Installation, DECA Aviation Engineering Limited document DOP 05295, Revision 2, dated 14/12/05 or later EASA approved revision</td>
<td>Data package containing the following documents, and additional supporting data: 1. Design Compliance Program for ASIRAS/Radar System Installation, DECA Aviation Engineering Limited document DOP 05295, Revision 12, Static Strength and Compliance Requirements, DECA Aviation Engineering Limited Engine...</td>
<td>1. DECA Aviation Flight Manual Supplement FM05205050, Revision 1, is required</td>
<td>18/07/2007</td>
<td>Active</td>
<td>1016830</td>
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<td>EASA STC no.</td>
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<td>Foreign/NAA STC no.</td>
<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>EASA.IM.A.S.01922</td>
<td>TIMCO AVIATION SERVICES INC.</td>
<td>REV 1</td>
<td>FAA STC ST0372AT</td>
<td>BOEING 767-300</td>
<td>Installation of Lower Lobe Crew Rest Module (CRM) and Main Deck/Vestibule in accordance with Timco Master data List 04T140401, Rev H, dated December 7, 2007 or later approved revisions approved by EASA in accordance with EASA ED Decision 2004/040/CF (or subsequent revisions of this decision).</td>
<td>1. Approved Aircraft Flight Manual Supplement TIM-AMM 01044 Rev A, dated December 14, 2007. Instructions for continued Airworthiness 04T104R007, Rev A, dated November 9, 2006.</td>
<td>1. This approval should not be extended to other Aircraft of this model on which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationship between this change and any other previously approved modifications will produce no adverse...</td>
<td>16/01/2008</td>
<td>Active</td>
<td>10018317</td>
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<tr>
<td>EASA.IM.A.S.01925</td>
<td>PRATT &amp; WHITNEY</td>
<td>REV 1</td>
<td>FAA STC ST0228BO</td>
<td>MD-88 DC-9-87 (MD-87) DC-9-93 (MD-83) DC-9-92 (MD-82) DC-9-81 (MD-81)</td>
<td>Installation of P&amp;W JTBD-200 Series Engines incorporating the improved forward Fan Case as described in the P&amp;W Service Bulletin JTBD 6478, Rev. 1, dated 1 August 2007.</td>
<td>FAA Service Bulletin JTBD 6478, Rev. 1, dated 1 August 2007.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description.&quot; Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other...</td>
<td>04/10/2007</td>
<td>Active</td>
<td>10018334</td>
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<tr>
<td>EASA.IM.A.S.01931</td>
<td>SATRAN CABIN INC.</td>
<td>FAA STC ST0185LA</td>
<td>BOEING 747-400</td>
<td>Installation of pantry</td>
<td>C &amp; D Zodiac Inc. drawing list C4812000 Revision F dated November 09, 2006 or later approved revision FAA STC 7071003LA</td>
<td>--</td>
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<td>19/12/2006</td>
<td>Active</td>
<td>10018317</td>
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<tr>
<td>EASA.IM.A.S.01932</td>
<td>AVIATION COMMUNICATION AND SURVEILLANCE SYSTEMS, L.L.C. (ACSS)</td>
<td>FAA STC SA4828NM</td>
<td>NOT CERT. 747-100B 747-SP 747-200F, 747-300, 747-400, 747-100, 747-200B, 747-200C</td>
<td>Installation of a Honeywell Traffic Alert and Collision Avoidance System (TCAS II) in accordance with FAA approved Honeywell Master Data List number ML4071596</td>
<td>Data package containing the following documents, and additional supporting data: 1. FAA Letter to Honeywell of approval pertaining to Honeywell previously approved STCs TCAS II installations to be upgraded to TCAS (10004) and to Mode S (10004) as indicated in Honeywell Master Drawing List for fol...</td>
<td>FAA approved Flight Manual Supplement required for this configuration is listed on the appropriate approved Honeywell and ACSS Master Index List.</td>
<td>31/07/2006</td>
<td>Active</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
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<td>Aircraft/Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>ST013685-C-D</td>
<td>GULFSTREAM AEROSPACE SERVICES CORP.</td>
<td>t/a GULFSTREAM AEROSPACE CORPORATION</td>
<td>t/a GULFSTREAM</td>
<td>Installation of Dual Universal Avioniks UNS-1C+ Flight Management Systems. FAA STC 0103685-C-D</td>
<td>AFMS CEA111B3005/CA</td>
<td>CEA111B4000</td>
<td>This STC is only applicable to A/C with previous installed dual Universal Avioniks UNS-1C FMS.</td>
<td>13/07/2006</td>
<td>Active</td>
<td>1001540</td>
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<tr>
<td>ST02989AT</td>
<td>STAR AVIATION. Inc.</td>
<td></td>
<td>CESSNA 550</td>
<td>The modification will consist of a composite structure within metal fuselage which will be fitted over the center of the existing Glareshield. An area of the existing Glareshield will be removed to allow for wires to be routed to the communication and switches which are relocated to the new Glareshield.</td>
<td></td>
<td></td>
<td>Master Data List R25-0283 Rev. E, dated 17 January 2006</td>
<td>12/06/2006</td>
<td>Active</td>
<td>1001560</td>
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<tr>
<td>ST06630AC</td>
<td>SIERRA INDUSTRIES Ltd.</td>
<td></td>
<td>CESSNA 501</td>
<td>Installation of Meggitt Engine Instrument Display System according to FAA STC ST06630AC.</td>
<td></td>
<td></td>
<td>Sierra Inc. Drawing S4500-001, Rev. A, dated 17.03.2006; or later FAA approved Revision for Cessna 501 equipped with Williams FJ44 engine FAA approved Sierra Inc. AMF-S Doc. No. 115-602, dated 29.03.12.</td>
<td>15/03/2007</td>
<td>Active</td>
<td>1001561</td>
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<tr>
<td>REIMS AVIATION</td>
<td>FAA STC SA03289W</td>
<td>LEARJET 36</td>
<td>Operation of airplane on leaded and unleaded automotive gasoline, 91 minimum antiknock index (RON + MON) / 2 per ASTM Specification D-439.</td>
<td>CESSNA F-172N</td>
<td>Petersen Drawing No. RF-172-91 dated January 18, 1995.</td>
<td>Petersen Drawing No. RF-172-91 dated January 18, 1995.</td>
<td>Limited to those airplanes equipped with gravity fuel feed to the carburetor and STC certified or approved engine installation. All engines must have separate approval for operation on automotive gasoline in the form of Supplemental or Type Certificate approval. The original Lycoming O-320-H2 and O-335-H2 ADs are removed.</td>
<td>24/04/2006</td>
<td>Active</td>
<td>1001563</td>
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<tr>
<td>CESSNA 172P</td>
<td>PETERSEN AVIATION, INC.</td>
<td>FAA STC SA2600CE</td>
<td>LEARJET 35</td>
<td>Operation of airplane on leaded and unleaded automotive gasoline, 91 minimum antiknock index (RON + MON) / 2 per ASTM Specification D-439.</td>
<td>CESSNA 172P</td>
<td>Petersen Drawing No. 172-91 dated February 13, or later approved revision. 2. Airplane Flight Manual Supplement dated March 30, 1990, or later approved revision for 172G, SerialNumbers 1727050 &amp; 17271059 thru 17274009 &amp; 172P Models only.</td>
<td>Limited to those airplanes equipped with gravity fuel feed wing fuel tanks as the main fuel supply to the carburetor and certified (original or STC) or approved engine installations. All engines must have separate approval for operation on automotive gasoline in the form of Supplemental or Type Certificate approval.</td>
<td>20/04/2006</td>
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<td>EASA STC no.</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.01960</td>
<td>GULFSTREAM AEROSPACE SERVICES CORP d/b/a, GULFSTREAM AEROSPACE CORPORATION d/b/a</td>
<td>FAA STC/No ST0237CH-D</td>
<td>GULFSTREAM ASTRON SPX GULFSTREAM 100 Installation of a Corporate Aircraft Completion in accordance with Gulfstream Master Drawing List GA120484000, Rev B dated July 19, 2006 or later approved revision</td>
<td>IFCA No.145384900</td>
<td>Limited to S/N 158</td>
<td>11/08/2006</td>
<td>Active</td>
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<td>EASA.IM.A.S.01961</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>FAA STC/No ST009158-E-D</td>
<td>BOEING 737-500 BOEING 737-400 BOEING 737-300 Installation of the Rosemount Aerospace Cockpit Door Surveillance system (CDSS).</td>
<td>1. Rosemount Aerospace Data List, Doc No. D003988, Revision B, dated February 15, 2006, or later approved revisions, 2 Rosemount Aerospace Doc No. D0030936, Revision A, dated December 16, 2005, or later approved revisions. 3 Rosemount Aerospace Instructions for Continued Airworthiness, Doc No. D053...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td></td>
<td>20/04/2006</td>
<td>Active</td>
<td>10015857</td>
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<td>EASA.IM.A.S.01962</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC C-LS06-020-D</td>
<td>BOMBARDIER BD-100-1A10 Aircraft completion, 10 Passenger interior configuration, in accordance with MDS E790000</td>
<td>AFM Supplement E790090</td>
<td>S/N 20079</td>
<td>24/04/2006</td>
<td>Active</td>
<td>10015858</td>
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<td>EASA.IM.A.S.01963</td>
<td>MTI POLYFAB Inc.</td>
<td>TCCA STC SA03-19</td>
<td>CL-600-1A11/2A12/2B16 Installation of Insulation Blankets</td>
<td></td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td></td>
<td>08/06/2006</td>
<td>Active</td>
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<td>EASA.IM.A.S.01967</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA STC SA03-43</td>
<td>CL-600-2B19 Installation of Safe Flight Autopower (Autoflight) System</td>
<td>Transport Canada STC SA03-43 Issue 1 dated May 09, 2003, MenConsulting Services Modification Data Summary D45150 Safe Flight AFM Supplement R-2827 Revision NC or later approved revision</td>
<td>--</td>
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<td>14/06/2006</td>
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<td>EASA.IM.A.S.01973</td>
<td>COLUMBIA AVIONICS INC.</td>
<td>FAA STC SA01156WI</td>
<td>CESSNA 551 CESSNA 501 Installation of dual RVSM compliant ISS &amp; J2 (Kollsman) air data system.</td>
<td>According to Columbia Avionics Master Document List Doc No. 30141024, Rev. 4 or later FAA approved Revision to be operated in accordance with either of FAA approved Columbia Avionics AFM-S Document No. 30141039, original issue, or later FAA approved Revisions or FAA approved Columbia Avionics...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 3. The installation of this modification by thee...</td>
<td></td>
<td>12/02/2009</td>
<td>Active</td>
<td>10015860</td>
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<tr>
<td>EASA.IM.A.S.01974</td>
<td>BEECHCRAFT CORPORATION REV. 2</td>
<td>FAA ST100066C-D</td>
<td>HAWKER 800XP, HAWKER 850XP, HAWKER 750, HAWKER 800XP Installation of B.F. Goodrich WX-1000E Stormscope.</td>
<td>1. In accordance with Master Drawing List 850-550 Revision G or later approved revision. 2. airplane Flight Manual Supplement 850-327 Revision RS or later approved revision. 3. Maintenance Manual Addenda: 34-43-10.</td>
<td></td>
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<td>20/05/2008</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<tr>
<td>EASA.IM.A.S.01975</td>
<td>BEECHCRAFT CORPORATION</td>
<td>REV. 1</td>
<td>FAA ST00865C</td>
<td>HAWKER 800XP</td>
<td>Installation of a Davtron Clock</td>
<td>Master Drawing List 800-553 Revision G or later approved revision</td>
<td>Aircraft Flight Manual Supplement 800-359 Revision R2 or later approved revision</td>
<td>Maintenance Manual Addenda 31-20-45</td>
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<td>BEECHCRAFT CORPORATION</td>
<td>REV. 2</td>
<td>FAA STC ST101085C-D</td>
<td>HAWKER 800XP</td>
<td>Installation of a Cockpit Speaker Volume Recalibration.</td>
<td>In accordance with : Master Drawing List 800-565 Revision IF or later approved revision</td>
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<td>FAA STC ST101065C-D</td>
<td>HAWKER 800XP</td>
<td>Installation of a Cockpit Speaker Inhibit Switch.</td>
<td>1. In accordance with : Master Drawing List 800-524 Revision L or later approved revision.2. Flight Manual Supplement 800-310 Revision R8 dated September 13, 2007 or later approved revision.3. Maintenance manual addenda 25-50-20.</td>
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<td>FAA ST101285C-D</td>
<td>HAWKER 800XP</td>
<td>Installation of a Precise Flight Pulselight System Activated by (existing) TCAS.</td>
<td>1. In accordance with : Master Drawing List 800-519 Revision L or later approved revision.2. Flight Manual Supplement 800-513 Rev R3 or later approved revision.3. Maintenance Manual Addenda 23-40-15.</td>
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<td>REV. 2</td>
<td>FAA ST101785C-D</td>
<td>HAWKER 800XP</td>
<td>Installation of a Satellite Communications System.</td>
<td>1. In accordance with : Master Drawing List 800-509 Revision J or later approved revision.2. Flight Manual Supplement 800-314 Revision R3 or later approved revision.3. Maintenance Manual Addenda 23-16-20.</td>
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<td>REV. 4</td>
<td>FAA ST06615C</td>
<td>HAWKER 800XP</td>
<td>Installation of Cabin Interior and Avionics Systems</td>
<td>In accordance with : Master Drawing List 800-181 Revision A/Y or later approved revision For Hawker 800XP, Hawker 850XP, Hawker 900XP: Aircraft Flight Manual Supplement 800-280 Revision R8 or later approved revision. For Hawker 750: Aircraft Flight Manual Supplement 800-350 Revision R3 or later approved revision.</td>
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<td>BEECHCRAFT CORPORATION</td>
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<td>FAA-SAB083SW</td>
<td>HAWKER 800XP HAWKER 800XP HAWKER 800XP HAWKER 800XP HAWKER 750, HAWKER 800 BAE 125 SERIES 800A</td>
<td>Installation of Cabin Headliner/Window Panel Brackets.</td>
<td>In accordance with Master Drawing List 800-116 Revision 2 dated May 16, 2006 or later approved revision.</td>
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<td>HAWKER 800XP HAWKER 800XP HAWKER 800XP HAWKER 800XP HAWKER 750, HAWKER 800 BAE 125 SERIES 800A</td>
<td>Installation of a Therapeutic Oxygen System.</td>
<td>1. In accordance with Master Drawing List 800-125 Revision B or later approved revision 2. Flight Manual Supplement 125-037 Rev 6 or later approved revision.</td>
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<td>FAA-SA7944SW</td>
<td>HAWKER 800XP HAWKER 800XP HAWKER 800XP HAWKER 800XP HAWKER 750, HAWKER 800 BAE 125 SERIES 800A</td>
<td>Installation of a Cabin Oxygen System.</td>
<td>1. In accordance with Master Drawing List 800-107 Revision D dated August 28, 2006 or later approved revision 0. Maintenance Manual Addenda: 35-06-00 &amp; 35-20-05</td>
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<td>BEECHCRAFT CORPORATION</td>
<td>REV. 1</td>
<td>FAA-SA3810SW-D</td>
<td>HAWKER 850XP HAWKER 850XP HAWKER 850XP HAWKER 850XP HAWKER 750, HAWKER 850 BAE 125 SERIES 800A</td>
<td>Installation of Precise Pulsatile Control Unit</td>
<td>1. In accordance with Master Drawing List 800-048 Revision Y or later approved revision 0. Flight Manual Supplement 800-231 Rev R103. Maintenance Manual Addenda: 33-40-05, 33-40-10</td>
<td>1. 103 Installation is for Models Hawker 800XP, Hawker 850XP and Hawker 900XP. All installations except 103 is for Models BAE. 125 Series 800A, Hawker 800, Hawker 800XP.</td>
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<td>BEECHCRAFT CORPORATION</td>
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<td>FAA-STO1342W-D</td>
<td>HAWKER 850XP HAWKER 850XP HAWKER 850XP HAWKER 850XP HAWKER 750, HAWKER 850 BAE 125 SERIES 800A</td>
<td>Installation of Collins Integral Flight Information System (IFIS 5000) and Proline 21 Radios.</td>
<td>1. In accordance with Master Data List AAC-0404-001 Rev B, or later approved revision 2. AFMS AAC-0406-001 RS, or later approved revision 3. Maintenance Manual Supplement AAC-0407-001 Rev - , or later approved revision.</td>
<td>1. Prior installation and approval of Kit 149-3419-0001 is a prerequisite; 2. Prior installation and approval of Kit 149-3430-0001 is a prerequisite;</td>
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<td>BEECHCRAFT CORPORATION</td>
<td>REV. 2</td>
<td>FAA-STO9772SC-D</td>
<td>HAWKER 900XP HAWKER 900XP HAWKER 900XP HAWKER 900XP HAWKER 750, HAWKER 800 BAE 125 SERIES 800A</td>
<td>Installation of a Smoke Detector System.</td>
<td>1. In accordance with Master Drawing List 800-192 Revision E or later approved revision 2. Airplane Flight Manual Supplement 801-297 Revision RS or later approved revision 3. Maintenance Manual Addendum: 26-11-00.</td>
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<td>EASA.IM.A.S.02002</td>
<td>LEARJET Inc.</td>
<td>BOMBARDIER CL-600-2B16 (604)</td>
<td>Installation of Oxygen System-Passenger &amp; crew (14500ft)</td>
<td>Learjet Inc. Master Document List DCSF3250500 Revision NC dated April 18, 2002 or later approved revision.AFM Supplement ref D-FM3250502 Revision NC dated March 12, 2002 or later approved revision.</td>
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<td>EASA.IM.A.S.02003</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA STC SA03-113</td>
<td>Installation of Complete Custom Aircraft Interior</td>
<td>Transport Canada STC SA03-13 Aero Consulting Services Ltd Configuration Definition List D817090 Revision GAero Consulting Services AFF Supplement D817090 Revision N/C or later approved revision</td>
<td>Limited to ASN 5591 only</td>
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<td>EASA.IM.A.S.02004</td>
<td>GODERICHER AIRCRAFT Inc.</td>
<td>O-LSA06-081/ D (LSTC)</td>
<td>Cabin Entertainment Equipment Installation.</td>
<td>Cabin Entertainment Equipment Installation according Master/Drawing List MOL06025 Rev. N/C, dated 30 March 2006 or later approved revision.Maintenance must be in accordance with Antoni Design Services Ltd. Instructions for Continued Airworthiness KIA9602-1 Rev. N/C, dated 27 March 2006 or later.</td>
<td>This Installation is limited to the Serial No. RK-37 (Registration SE-DRS).This approval is for electrical/avionics aspects only. Structural aspects of this installation are not included and require separate approval.</td>
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<td>EASA.IM.A.S.02005</td>
<td>GODERICHER AIRCRAFT Inc.</td>
<td>O-LSA06-117/ D (LSTC)</td>
<td>Change in interior seating arrangement, addition of sink boasted re-lavatory, fabrication and installation of RH fwd cabinet, revision of existing LH fwd cabinet, revision trimantor finishing details, and addition of new fuselage antennae</td>
<td>Installation according Phoenix Aero Aviation Engineering Ltd. Engineering Service Bulletin FA-ESB-06-022 Issue 1, Rev. Original, approved 28 April 2006 or later approved revision.Operation must be in accordance with Phoenix Aero Aviation Engineering Ltd. Flight Manual Supplement FMS-PA-06-003 Issue 1..</td>
<td>1. This Installation is limited to the Serial No. RK-37 (Registration SE-DRS).2. Compatibility of this modification with other previously installed modifications must be determined by the installer.</td>
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<td>EASA.IM.A.S.02007</td>
<td>AVIATION FABRICATORS, Inc.</td>
<td>650</td>
<td>Installation of a 1, 2 or 3 place side facing Divan</td>
<td>Installation in accordance with Aviation Fabricators STC Data Package AF-263, Rev K, dated June 9, 2004, with an approval date of July 28, 2004, or later approved revision.</td>
<td>Master Data List ECS-202803, revision G, dated 12 April 2006 or later approved revision</td>
<td>Applicable to aircraft serials 650-0023 and subsequent. This STC approves only the installation of the parts contained in the installation documents referenced above. The requirements for aisle width, passenger oxygen systems, seats and safety belts, and upholstery coverings will require a separate EA...</td>
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<td>EASA.IM.A.S.02008</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC STC02075CH</td>
<td>AIRBUS A300 MODEL B4-2C AIRBUS A300 MODEL B4-203 AIRBUS A300 MODEL B4-103 AIRBUS A300 MODEL B2K-3C AIRBUS A300 MODEL B2-203 AIRBUS A300 MODEL B2-1C AIRBUS A300 MODEL B2-1A</td>
<td>Installation of Universal Aeronics Systems Corporation Terrain Awareness and warning System (TAWS)</td>
<td>Aircraft Flight Manual Supplement ECS 202804, revision B, dated 11 October 2005 or later approved revision/instruction for continued airworthiness ECS 202804, revision B, dated 5 December 2005. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall...</td>
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<td>EASA.IM.A.S.02009</td>
<td>EMBRAER - EMPRESA BRASILERADE AERONAUTICA S.A. - PC 17B4 GEC</td>
<td>CTA STC02006502-09</td>
<td>EMBRAER EMB-135BJ</td>
<td>Installation of a 32&quot; LCD monitor and PlayStation Entertainment System in Avionic Services master drawing list 135-23-001 Rev. D dated 28 Nov. 05 or later approved revision</td>
<td>CTA STC02006502-09</td>
<td>1. Limited to EMB-135BJ SN 1450008 2. Instructions for Continued Airworthiness Report No. 135-23-0008 Rev. 0 dated 9 Sep. 05 or later accepted revision 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot;...</td>
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<td>WIPARE Inc.</td>
<td>FAA STC SAI000SL</td>
<td>DHC-2, MK1</td>
<td>Installation of Wipaire Model 6000 or 8150 Seaplane or Amphibious Floats</td>
<td>In accordance with Wipaire Master Drawing List ref WD-MD115, dated June 3 1982, or later approved revisions. FAA Approved Flight Manual Supplement dated June 3, 1982 is required for Amphibious float installations, or later approved revisions. FAA Approved Flight Manual Supplement dated April 1, 1983.</td>
<td>FAA Approved Flight Manual Supplement dated April 1, 1983...</td>
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<td>ELLIOTT AVIATION</td>
<td>FAA STC SAI2264CH</td>
<td>BEECH B300C BEECH B300 BEECH B300CT BEECH B300C BEECH B300 BEECH 300</td>
<td>Airworthiness certification of Reduced Vertical Separation (RVSM) and installation of Air Data Systems</td>
<td>Installation in accordance with Elliott Aviation Master Drawing List, Report No MDL-2468-00. Revision C, dated April 16, 2006 or later approved revision</td>
<td>None</td>
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<td>COMMUTER AIR TECHNOLOGY INC.</td>
<td>REV. 1</td>
<td>FAA STC SA00104A</td>
<td>BEECH B200 T BEECH B200 CT BEECH B200 C BEECH B200 BEECH A200 CT BEECH A200 BEECH 2007 BEECH 200 CT BEECH 200 BEECH 200</td>
<td>Installation of Alternate Main Landing Gear Tires, Michelin®/N 263-335-0. Revision 1 issued to add models B200GT and B200CCT.</td>
<td>- Installation in accordance with Commuter Air Technology Top Drawing LE10000 Revision NC, dated May 3, 1996 or later approved revisions. - Operation in accordance with Commuter Air Technology Approved Flight Manual Supplement LE-10001, dated May 14, 1996 or later approved revisions.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>TELEDYNE TECHNOLOGIES INCORPORATED</td>
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<td>SA10169O</td>
<td>BEECH B300 C BEECH B200 T BEECH B200 CT BEECH 300LW BEECH 300 BEECH 2007 BEECH 200 CT BEECH 200 BEECH 200</td>
<td>Replace existing nickel-cadmium battery with Teledyne G-6381E lead-acid battery or with Teledyne G-6381ES Sealed Recombinant lead-acid storage battery</td>
<td>Installation instructions for replacing existing nickel-cadmium battery with Teledyne G-6381E lead-acid battery, dated August 3, 1978, or later approved revisions.Teledyne Master Data List 1016-7, Revision &quot;NC&quot;, dated December 13, 2000, or later approved revisions.FAA Approved Teledyne Battery Product Airplane Flight Manual Supplement, dated August 3, 1978, for G-6381ES battery, or later approved revisions.</td>
<td>FAA approved Teledyne Battery Product Airplane Flight Manual Supplement, dated August 3, 1978, for G-6381ES battery, or March 21, 2001, for G-6381ES battery, or later approved revisions is required as part of this installation.</td>
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<td>EASA.IM.A.S.02018</td>
<td>TELEDYNE TECHNOLOGIES INCORPORATED</td>
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<td>SA10179O</td>
<td>BEECH E90 BEECH C90 BEECH B99 BEECH A100 BEECH 65-A60-1 BEECH 65-A60 BEECH 65-50</td>
<td>Replace existing nickel-cadmium battery with Teledyne Battery Model No. G-638E, G-6381E or G-6381ES</td>
<td>Teledyne Battery Installation Instruction No. SA 10175O, Revision A, dated April 7, 1998, or later approved revision.</td>
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### List of EASA Supplemental Type Certificates

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<td>EASA.IM.A.S.02036</td>
<td>HAWKER BEECHCRAFT CORPORATION (HBC)</td>
<td>Installation of a Logo Light on the lower surface of the Horizontal Stabiliser</td>
<td>Installed in accordance with DevOre Top Drawing No 1250003, dated october 30, 1998 or later approved revision</td>
<td>Data package containing the following documents and additional supporting data: 1. Certification Plan Boeing 747-400 Series Elementary &amp; Enhanced Surveillance Program, Cascade Aerospace Engineering document PR 1-3710.1.1. Rev. Nil, or later approved revisions. 2. Boeing 747-400 Series Elementary...</td>
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<td>TIMCO AVIATION SERVICES INC.</td>
<td>Interior Reconfiguration</td>
<td>- TIMCO Master Data List 04T110010 Revision C, dated May 31/2005 or later approved revisions - TIMCO Instructions for Continued Airworthiness 04T110R115 Revision IR dated May 12 2006</td>
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<td>EASA.IM.A.S.02041</td>
<td>BOMBARDIER INC.</td>
<td>Aircraft completion, 10 passenger interior configuration, in accordance with MDS F0260000 (Original STC number TransportCanada STC C-LSA06-270/D)</td>
<td>Applied Supplement F029090 Limited to S/N 20106</td>
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<td>EASA.IM.A.S.02042</td>
<td>MILLENNIUM CONCEPTS, INC.</td>
<td>Installation of Exterior Inter in Cessna Models 208, 208B Caravan</td>
<td>In accordance with Millenium Concepts Inc Master Drawing list (MDL) 2015-MDL22. Rev Y dated May 15 2007, or later approved revisions.</td>
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<td>EASA.IM.A.S.02044</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>Installation of Honeywell MCS-7000 SATCOM system</td>
<td>In accordance with Gulfstream Aerospace Corporation Index List GC51 1038013, Revision B, dated January 17, 2005 or later approved revision. FAA approved Flight Manual Supplement, GC51 1038004, Revision NC, dated January 20, 2005 or later approved revision.</td>
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<td>EASA.IM.A.S.02048</td>
<td>WESTERN AVIONICS</td>
<td>Installation of Honeywell EGPAWS Mark V/III computer, air data computer, circuit breakers, annunciators, flap positionswitch, D/AInput, GPS antenna, cockpit speaker and associated wiring according Transport Canada Supplemental Type Certificate SA04-26, Issued, March 09, 2005</td>
<td>1. Installation is to be done in accordance with Transport Canada approved Western Avionics Engineering Inter 92350-401, Revision 4, dated 31 January 2005, or later Transport Canada approved revision. 2. Transport Canada approved Western Avionics Flight Manual Supplement 92350-801, Revision 3, dated...</td>
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<td>EASA.IM.A.S.02049</td>
<td>DUNCAN AVIATION, INC.</td>
<td>Installation of Honeywell MK II Enhanced Ground Proximity Warning System (EGPWS) to software version 226-226 and Optional RAAS</td>
<td>1. Only Software standard -226-226 or later is approved for installation. 2. Prior to installation of this modification the installer must determine that the internrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the air...</td>
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<td>SA01-98</td>
<td>CASCADE AEROSPACE Inc.</td>
<td>FAA STC</td>
<td>APU Oil Cooler Door Deletion per TCCA STC SA01-98, Issue 1, Issue Date: 23 August 2001</td>
<td>FAA STC ST0273LA_DMaster List</td>
<td>Modification of an aircraft by this STC obligates the aircraft operator to include the maintenance information TE-55325000-IFCA in the operators Maintenance Manual &amp; the Operators/Scheduled Maintenance Program. The equipment for which provisions were procured has not been certified. EASA approval will...</td>
<td>03/11/2006</td>
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<td>SA023816</td>
<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>Installation of a Complete Custom Interior</td>
<td>FAA STC ST02123A_DMaster List</td>
<td>DC35325000 Revision D dated May 23 2003 or after approved revision</td>
<td>02/07/2006</td>
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<td>SA026256</td>
<td>LEARJET Inc.</td>
<td>FAA STC</td>
<td>Installation of Honeywell Enhanced Ground Proximity Warning System with Runway Awareness &amp; Advisory System (RAAS)</td>
<td>FAA STC SA8109NM-D</td>
<td>Master Drawing List CA-1072-TUC Revision AC or after approved revision. Learjet Inc. AFM Supplement BAS03047012 Revision NC or after approved revision.</td>
<td>05/07/2007</td>
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<td>SA038626</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC</td>
<td>Installation of Collins Transponder with Flight ID for Elementary Surveillance (-501 configuration) or Enhanced Surveillance (-503 configuration).</td>
<td>AFM CEA1126M000 / IFCA CEA1126A0005/03</td>
<td>1. The aircraft is limited to Private use only (not to be operated for hire, reward or common carriage). 2. This STC is approved only for the product configuration as defined in the above Master Document List. 3. Limitations and Conditions specified in FAA STC ST10593SC remain applicable. 4. Prior to installation of the above...</td>
<td>24/07/2006</td>
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<tr>
<td>SA038629</td>
<td>ASSOCIATED AIR CENTER, L.P.</td>
<td>FAA STC</td>
<td>Installation of an Executive Style Interior (reference FAA STC ST10593SC dated 6 February 2007). First installation is on M550-355/Note: This STC superseded EASA...</td>
<td>Master Document List, Document Number 23B8000000, Revision A, dated 5 February 2007, or after EASA approved revision. - Instructions for Continued Airworthiness, including Amendment Limitations Section, Document Number 2388000000, Revision B, dated 5 February 2007, or after EASA approved revision...</td>
<td>Configuration 1 DHC-8 Series 100/200/300 aircraft modified in accordance with STC FAA/EASA/MA. S. 01714Configuration 2 DHC-8 Series 100/200/300 aircraft in accordance with STC FAA/EASA/MA. S. 00221 Equivalent to Bombardier Model 8328 and...</td>
<td>14/09/2009</td>
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List of EASA Supplemental Type Certificates

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<tr>
<td>CASCADE AEROSPACE Inc.</td>
<td></td>
<td>STC03294AT-D</td>
<td>GULFSTREAM G-V</td>
<td>Completion and installation of the custom interior on Aircraft Serial 5071</td>
<td>In accordance with Gulfstream Aerospace Corporation Master Index List GC1491410051, Revision D signed August 31, 2000. FAA approved revision.</td>
<td>Aircraft Serial 5071 only</td>
<td>21/08/2006</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>STC03284AT-D</td>
<td>GULFSTREAM G-VX</td>
<td>Completion and installation of the custom interior on Aircraft Serial Number 4045.</td>
<td>In accordance with Gulfstream Aerospace Corporation Master Index List GC1491394065, Revision D dated 30 August 2006, or later approved revision.</td>
<td>Aircraft Serial Number 4045.</td>
<td>27/10/2006</td>
<td>Active</td>
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<td>AAR AIRCRAFT SERVICES INC.</td>
<td>SAT467SW</td>
<td>F27 (MARK TBD)</td>
<td>Installation of dual layer air conditioning systems in accordance with AAR Oklahoma, Inc., Drawing List 19U0002, Revision F, dated July 19, 1988, or later FAA approved revision.</td>
<td>AAR Oklahoma, Inc., Drawing List 19U0002, Revision F, dated July 19, 1988, or later FAA approved revision.</td>
<td>1. FAA approved flight manual supplement dated August 10, 1998, is required. 2. Compatibility of this modification with previously installed equipment must be determined by installer.</td>
<td>13/10/2006</td>
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<td>BOMBARDIER AEROSPACE</td>
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<td>TCCA STC C-LSA06-3040</td>
<td>CL-600-2B16</td>
<td>Aircraft Completion</td>
<td>Transport Canada Limited STC C-LSA06-3040.</td>
<td>Restricted to ASN 5648</td>
<td>16/11/2006</td>
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<td>EASA.IM.A.S.02071</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA STC C-LSA08-243-D ISSUE 1</td>
<td>BIJONARTER CL-600-2B16 (604)</td>
<td>Aircraft Completion</td>
<td>Transport Canada STC C-LSA08-243-D Issue 1. Aero Consulting Services Modification data Summary E933000 Revision NC or later approved revision. Aero Consulting Services AFM Supplement E963000 Revision NC or later approved revision.</td>
<td>1. Restricted to ASN 5444.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>19/02/2006</td>
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<td>EASA.IM.A.S.02072</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>FAA STC ST00255DE-D</td>
<td>BA1 146/AVRO 146 SERIES</td>
<td>Installation of Allied Sigint VHF AFIS</td>
<td>FAA STC ST00255DE-D Learjet Inc Master Document List BAS4601001 Revision B dated May 30, 1997 or later approved revision</td>
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<td>EASA.IM.A.S.02075</td>
<td>MID-CANADA MOD CENTER</td>
<td>SAJ-92</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Cockpit voice recorder replacement with Universal CVR 30 or CVR 120</td>
<td>Avionics Design Services Ltd. Master Drawing List ref. M80374 revision B, dated August 6, 2004</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>14/12/2006</td>
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<td>EASA.IM.A.S.02077</td>
<td>GROVE AIRCRAFT LANDING GEAR SYSTEMS INC.</td>
<td>SAGI704A</td>
<td>SEE APPROVED MODEL LIST</td>
<td>Installation of the Grove Disk Brake System in accordance with FAA approved Master Drawing List Number MDL2337-00001, Revision B, dated December 8, 2004, or later FAA approved revision</td>
<td>FAA approved Master Drawing List Number MDL2337-00001, Revision B, dated December 8, 2004, or later FAA approved revision</td>
<td>The installation applies only to aircraft that have hydraulic brake systems installed. The installation should not be incorporated in any aircraft unless it is determined that the interrelationship between this installation and any ant previous installation</td>
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<td>EASA.IM.A.S.02078</td>
<td>DEVORE AVIATION CORPORATION OF AMERICA</td>
<td>FAA STC SA74735SW REV 1</td>
<td>BOEING 757</td>
<td>Installation of tail floodlights in accordance with DeVore Drawing No. 37500001, Rev B, dated September 14, 1988, or later FAA approved revision.</td>
<td>Modification FAA STC No. SA74735SW</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST01673AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of a Honeywell P-880 Radar System</td>
<td>Installed in accordance with Gulfstream Aerospace Corporation Index List GCS1 1258001, Revision 02, dated June 20, 1997, or later approved revisions FAA approved Flight Manual Supplement, GCS1 1258001, Revision NC, dated June 30, 1997 or later approved revision is required. Continued airworthiness in...</td>
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<td>EASA.IM.A.S.02082</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST01673AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of a Transponder Flight Identification</td>
<td>Installed in accordance with Gulfstream Aerospace Corporation Index List GCS1 1268000, Revision NC, dated June 24, 2004, or later approved revisions FAA approved Flight Manual Supplement, GCS1 1268000, Revision NC, dated July 12, 2004 or later approved revision is required for the 501 configuration...</td>
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<td>EASA.IM.A.S.02083</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of a Vacuum Toilet System</td>
<td>Installed in accordance with Gulfstream Aerospace Corporation Index List GCS2 404002, Revision 02, dated July 02, 1997, or later approved revisions. Continued airworthiness instructions as defined in GCS2 400401, Revision NC, dated July 11, 1997, or later approved revision is required.</td>
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<td>EASA.IM.A.S.02084</td>
<td>DENEL OPTRONICSA DIVISION OF DENEL PTY LTD</td>
<td>CESSNA 208 AND 208B</td>
<td>Installation of Airborne Observation System</td>
<td>1. Max operating speed 155 kts IAS. 2. Observation system must be switched OFF during take-off and landing. 3. Persons on board during flight: limited to flight crew and operator(s) of observation system.</td>
<td>-</td>
<td>1. Installation of Collins Mode S Transponder, per STC SA2636CE (EASA.IM.A.S. 02087) is a prerequisite to this modification.</td>
<td>28/07/2006</td>
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<td>EASA.IM.A.S.02085</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>BOEING 747-500, 747-400, 747-300, 747-200F, 747-200C, 747-200B</td>
<td>Installation of Collins traffic alert and collision avoidance system (TCAS II)</td>
<td>Drawing Indexes 992-1978-001, 992-1989-001, and 992-2696-001. Aircraft Flight Manual Supplement ACC-99-042/-43/-44/-45.</td>
<td>-</td>
<td>1. Installation of Collins Mode S Transponder, per STC SA2636CE (EASA.IM.A.S. 02087) is a prerequisite to this modification.</td>
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<td>EASA.IM.A.S.02086</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>BOEING 747-500, 747-400, 747-300, 747-200F, 747-200C, 747-200B</td>
<td>Installation of Collins traffic alert and collision avoidance system (TCAS II)</td>
<td>Drawing Indexes 992-1978-001, 992-1989-001, and 992-2696-001. Aircraft Flight Manual Supplement ACC-99-042/-43/-44/-45.</td>
<td>-</td>
<td>1. Installation of Collins Mode S Transponder, per STC SA2636CE (EASA.IM.A.S. 02087) is a prerequisite to this modification.</td>
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<td>EASA.IM.A.S.02087</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>BOEING 747-500, 747-400, 747-300, 747-200F, 747-200C, 747-200B</td>
<td>Installation of Engine Fuel Drain Recirculation System</td>
<td>FAA STC SA996EADV - Master Drawing List CA-1077-TUC Revision C dated April 21, 1975 or later approved revision.</td>
<td>-</td>
<td>1. Compatibility of this design change with other previously approved modifications must be determined by the installer.</td>
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<td>EASA.IM.A.S.02088</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>BOEING 737-900, 737-800, 737-700, 737-600</td>
<td>Installation of the Mark II Aircraft Communications Addressing and Reporting System (ACARS) in the Boeing 737-600, -700, -800, -900 Series Aircraft.</td>
<td>MDL 756-0124-001, Revision K, dated March 24, 2006, or later EASA approved revisions.</td>
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<td>EASA.IM.A.S.02089</td>
<td>DEVOIRE AVIATION CORPORATION OF AMERICA</td>
<td>BOMBARDIER DHC-6 SERIES 300, 200, 100</td>
<td>Installation of Engine Fuel Drain Recirculation System</td>
<td>FAA STC SA996EADV - Master Drawing List CA-1077-TUC Revision C dated April 21, 1975 or later approved revision.</td>
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<td>EASA.IM.A.S.02090</td>
<td>LEARJET Inc.</td>
<td>BOMBARDIER CL-600-2B16 (604)</td>
<td>Installation of Ultra Electronics Active Noise &amp; Vibration Control system</td>
<td>FAA STC SA5129NM-D Master Drawing List CA-1077-TUC Revision C dated October 08, 1998 or later approved revision.</td>
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<td>EASA.IM.A.S.02091</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>FAA STD ST01424CH-D</td>
<td>Hawker 125 -650P SERIES, Hawker 125 -660 SERIES</td>
<td>Installation of Honeywell N1 Digital Electronic Engine Control (DEEC)</td>
<td>FAA STD ST01424CH-D</td>
<td>Garrett Aviation Services Drawing List 43-80027-001 Revision C dated May 11 2006 or later approved revision AFM Supplement 47-8027-001 Revision B dated May 11 2006 or later approved revision For Hawker 800XP/500XP AFM Supplement 43-8027-003 Revision B dated May 11 2006 or later...</td>
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<tr>
<td>EASA.IM.A.S.02092</td>
<td>AIRCRAFT SYSTEMS AND MANUFACTURING G,INC</td>
<td>ST10084SC</td>
<td>737-100, 737-200, 737-300</td>
<td>Installation of an Innovative Solutions &amp; Support (ISS) digital air data system consisting of an electric altimeter, Digital Air Data Computer (DADC), Air Data Display Unit (ADDU), and an attitude alert in accordance with ASM Master Data List 5572-0001.00, Revision C, Dated October 2, 2002 or lat...</td>
<td>- ASM Master Data List 5572-0001.00, Revision C, Dated October 2, 2002 or later EASA approved revision. - Airplane Flight Manual Supplement No 5572-21600-00, Revision IR, Dated October 17, 2002, or later EASA Approved Revision</td>
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<td>EASA.IM.A.S.02094</td>
<td>BARBARA AND ROBERT WILLIAMS</td>
<td>SAS72CE</td>
<td>A150L, A150K, 150M, 150L, 150K, 150JH, 150F, 150E, 150D, 150C, 150B, 150A, 150</td>
<td>Replacement of Continental O-200A with Lycoming 150 or 160hp Model O-320 engines</td>
<td>Installation, operation and Maintenance in accordance with the documents and data contained in FAA STC SAS72CE at latest approved revision</td>
<td>As defined in FAA STC SAS72CE Modified engine mounting structure and relocated battery installation must be inspected every 500 flight hours or annually, whichever occurs earlier for any cracking or corrosion and if found rectified with an EASA approved repair scheme</td>
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<td>EASA.IM.A.S.02095</td>
<td>ANDRA INTERNATIONAL LLC</td>
<td>FAA STC ST01926LA</td>
<td>747-400</td>
<td>Installation of Main Deck Cargo Loading System</td>
<td>FAA STC ST01926LA:Andra International Controlled Data List DL84400-10-4 Revision A dated October 09 2006 or later approved revision.</td>
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<td>12/01/2007</td>
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<td>EASA.IM.A.S.02098</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>C-LSA06-256/ D</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Aircraft Completion, 13 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA06-256/D to S/N 9188MDL. E76000 Rev NC or later TCCA approved revisions</td>
<td>SoC: C-06-0642, dated Sep 28, 2008FM: FM25000140 Rev NC or later TCCA approved revision/IFCA: GC91880000-IFCA</td>
<td>Limited to Aircraft Serial Number 262 only.</td>
<td>04/10/2006</td>
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<td>EASA.IM.A.S.02100</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TC-A STC C-LSA06-291/D</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Aircraft Completion, 12 Passenger Interior Configuration/Validation of TC-A Limited STC C-LSA06-291/D to S/N 90200/DL. F09000 Rev NC or later approved revisions</td>
<td>SoC: C-06-0917 dated Nov 24, 2006,FMS: G-FM2500014 Rev NC or later approved revisions,FCA: GC92000000-FCA</td>
<td></td>
<td>Refer to the Limitations sections of the required FM Supplement/WLAN system is compatible only to IEEE 802.11 (b) The use of WLAN is subject to operational rules.</td>
<td>30/11/2006</td>
<td>Active</td>
<td>10115557</td>
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<tr>
<td>EASA.IM.A.S.02101</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TC-A STC C-LSA06-336/D</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Aircraft Completion, 13 Passenger Interior Configuration/Validation of TC-A Limited STC C-LSA06-336/D to S/N 92000/DL. F06200 Rev NC or later approved revisions</td>
<td>SoC: C-06-0949, dated Dec 18, 2006,FMS: G-FM25000147 Rev NC or later approved revisions,FCA: GC92000000-FCA</td>
<td></td>
<td>Refer to the Limitations sections of the required FM Supplement/WLAN system is compatible only to IEEE 802.11 (b) The use of WLAN is subject to operational rules.</td>
<td>05/12/2006</td>
<td>Active</td>
<td>10115559</td>
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<td>EASA.IM.A.S.02102</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>C-LSA06-321/D</td>
<td>BOMBARDIER BD-700-1A0</td>
<td>Aircraft Completion, 14 Passenger Interior Configuration/Validation of TC-A Limited STC C-LSA06-321/D to S/N 92000/DL. F09000 Rev NC or later approved revisions</td>
<td>SoC: C-06-0954, dated Dec 18, 2006,FMS: G-FM25000147 Rev NC or later approved revisions,FCA: GC92000000-FCA</td>
<td></td>
<td>Refer to the Limitations sections of the required FM Supplement/WLAN system is compatible only to IEEE 802.11 (b) The use of WLAN is subject to operational rules.</td>
<td>20/12/2006</td>
<td>Active</td>
<td>10115559</td>
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<td>EASA.IM.A.S.02103</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TC-A STC C-LSA07-015/D</td>
<td>BOMBARDIER BD-700-1A0</td>
<td>Customized Interior Completion - SN 9195 Validation of TC-C STC C-LSA07-015/D.</td>
<td>AFM Supplement G-FM25000149 TC-A SoC letters C-06-0667 dated November 09, 2007 Instructions for Continuous Airworthiness : GC91960000-FCA</td>
<td></td>
<td>Refer to the Limitations sections of the required FM Supplement/WLAN system is compatible only to IEEE 802.11 (b) The use of WLAN is subject to operational rules.</td>
<td>29/11/2007</td>
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<td>EASA.IM.A.S.02106</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TC-A STC C-LSA07-013/D</td>
<td>BOMBARDIER BD-700-1A0</td>
<td>Aircraft Completion, 13 Passenger Interior Configuration/Validation of TC-A Limited STC C-LSA07-013/D to S/N 90200/DL. F09000 Rev NC or later approved revisions</td>
<td>SoC: C-06-0413 E786 dated Mar 01, 2007,FMS: G-FM25000151 Rev NC or later approved revisions,FCA: GC92000000-FCA</td>
<td></td>
<td>Refer to the Limitations sections of the required FM Supplement/WLAN system is compatible only to IEEE 802.11 (b) The use of WLAN is subject to operational rules.</td>
<td>21/03/2007</td>
<td>Active</td>
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<td>EASA.IM.A.S.02110</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>FAA STC ST08644SE</td>
<td>BAE 146/AVRO 146 SERIES</td>
<td>Installation of the Flight Data Acquisition Management System (FDAMS)</td>
<td>No change to existing documents</td>
<td></td>
<td>This change is not JAR-OPS-1.715 (EUROCAE ED55) compliant in the following parameters: Parameter 6: Pitch Attitude is only sampled once per second. Parameter 7: Roll Attitude is only sampled once per second. Parameter 23: Ground Spoiler position. Brake Selection is only sampled once per second with a R...</td>
<td>13/12/2006</td>
<td>Active</td>
<td>10115562</td>
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<td>EASA.IM.A.S.02113</td>
<td>AIRCRAFT SYSTEMS AND MANUFACTURING Q,INC</td>
<td>ST097445SC</td>
<td>BAE 146/AVRO 146 SERIES</td>
<td>Installation of a Digital Flight Data Recorder (DFDR) Expanded Parameter Sensor Upgrade on a British Aerospace Avro 146/Avro RJ85 Aircraft</td>
<td>ASM Top: Drawing List 5561-0001-00, Revision D, dated July 10, 2000, or later FAA approved revision</td>
<td></td>
<td>None</td>
<td>19/07/2006</td>
<td>Active</td>
<td>10115564</td>
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<tr>
<td>EASA.IM.A.S.02114</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST096395SC</td>
<td>GULFSTREAM 200</td>
<td>Installation of an Allied Signal DFDR i.e. MDS30-11000007</td>
<td>AFMS 30-1120011</td>
<td></td>
<td>None</td>
<td>16/10/2006</td>
<td>Active</td>
<td>10115565</td>
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<tr>
<td>EASA.IM.A.S.02115</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST094565C</td>
<td>GULFSTREAM 200</td>
<td>Installation of Lead Acid Batteries i.e. MDS30-1100014</td>
<td>AFMS 30-1120013</td>
<td></td>
<td>None</td>
<td>16/10/2006</td>
<td>Active</td>
<td>10115566</td>
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<tr>
<td>EASA.IM.A.S.02118</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TC-C STC C-LSA06-317/D</td>
<td>BOMBARDIER BD-700-1A0</td>
<td>Aircraft Completion, 14 Passenger Interior Configuration/Validation of TC-C Limited STC C-LSA06-317/D to S/N 9197MDL. F017000 Rev NC or later approved revisions</td>
<td>SoC: dated Jan 31, 2007,FMS: G-FM25000146 Rev NC or later TCCA approved revision,FCA: GC91970000-FCA</td>
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<td>Refer to the Limitations sections of the required FM Supplement/WLAN system is compatible only to IEEE 802.11 (b) The use of WLAN is subject to operational rules.</td>
<td>31/01/2007</td>
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<td>STC Holder</td>
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<td>Model</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>CIRRUS DESIGN CORPORATION</td>
<td>SA13595WI-D</td>
<td>CIRRUS SR20</td>
<td></td>
<td>Validation of FAA STC SA13595WI-D – “Installation of Bendix/King K887 ADF and/or RN 62A DME”</td>
<td>Installation in accordance with Master Drawing List 159-06310-001, Rev. C, or later FAA approved Revision</td>
<td>1. The installer must determine, if this design is compatible with previously approved modifications. If the holder agrees to permit another person to use this certification on another product, the holder must give the other person written evidence of that permission.2. This STC is approved on...</td>
<td>15/05/2007</td>
<td>Active</td>
<td>1015902</td>
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<tr>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST0306AT-D</td>
<td>GULFSTREAM GV-SF</td>
<td></td>
<td>Completion and installation of the custom interior on Aircraft Serial 5116</td>
<td>In accordance with Gulfstream Aerospace Corporation Master Index Lot: GCS1 4919115, Revision D, dated October 22, 2006 or later approved revisions; FAA approved Flight Manual Supplement, GC51 504#0001, Revision NC, dated December 22, 2004, or later approved revisions.</td>
<td>Aircraft Serial 5116 Only. The following STCs are required to be installed as part of this STC: FAA STC ST0196AT-D, EASA Approved ref: EASA.IM.A.S.01399, FAA STC ST0299AT-D, EASA Approved ref: EASA.IM.A.S.01399, FAA STC ST0298AT-D, EASA Approved ref: EASA.IM.A.S.01399, FAA STC</td>
<td>06/11/2006</td>
<td>Active</td>
<td>1015903</td>
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<tr>
<td>FIRST AIR MAINTENANCE SERVICES</td>
<td>SA06-39</td>
<td>AEROSPATIALE A320-200</td>
<td></td>
<td>Installation of Dual SAGEM F227 APRIS Attitude and Heading Reference System</td>
<td>Installation must be in accordance with First Air Master Data List, MDL-06-01, Revision A, dated 20th November 2006 or later approved revisions. Operation must be in accordance with First Air Flight Manual Supplement AFMS 06-004 Issue 1, 13th March 2006 or later approved revisions. Maintenance must be...</td>
<td>This STC is approved only for the aircraft models: ATR 42-300 / 320 / 400 / 500. ATR 42-200 / 300 / 400 / 500 ATR 72-101 / 102 / 201 / 202 / 211 / 212 / 213. ATR 42-320.</td>
<td>15/01/2007</td>
<td>Active</td>
<td>1015907</td>
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<td>L3 AVIATION PRODUCTS, INC</td>
<td>FAA STC ST02111AT</td>
<td>F28 MARK 0100</td>
<td></td>
<td>Flight Systems Engineering Report for the Additional Parameters for Flight Recording System on Fokker, F-100 Aircraft. L3 Communications Document No. FSE 58-06-01, Revision C.</td>
<td></td>
<td>1. n/a</td>
<td>20/12/2006</td>
<td>Active</td>
<td>1015911</td>
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<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC ST01159WI-D</td>
<td>LEARJET 36A</td>
<td></td>
<td>Installation of Honeywell IFR-VIII EGPWS FAA STC ST0115999-D</td>
<td></td>
<td>Certification Plan 06042019, rev. AFM Supplement 40067000.rev. A</td>
<td>None.</td>
<td>06/07/2007</td>
<td>Active</td>
<td>1015972</td>
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<td>AMTECH AERONAUTICAL LIMITED</td>
<td>SA02-132, ISSUE NO.3</td>
<td>382G</td>
<td></td>
<td>Oil Dispersant Spraying System Installation on Lockheed 328G aircraft.</td>
<td></td>
<td>Operation in accordance with Airplane Flight Manual Supplement FMS243001 Rev. 4, dated November 4, 2005. TCRA approved revision in conjunction with TCRA Operating Manual OP2439 Rev. 4 dated November 4, 2005 or later TCRA approved revision.</td>
<td>01/02/2007</td>
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<td>TIMCO AVIATION SERVICES INC.</td>
<td>FAA STC ST03030AT-D</td>
<td>AIRBUS A321</td>
<td></td>
<td>Installation of a doghouse (storage container) in the bulk cargo compartment.</td>
<td>Master Data List, Document Number 05T101L001, Revision a, dated 26 September 2006, or later EASA approved revision; Instructions for Continued Airworthiness, Document Number 05T101R008, Revision III, dated 09 September 2006, or later EASA approved revision.</td>
<td>Limitations and Conditions specified in FAA STC ST03030AT-0111R008 applicable.</td>
<td>18/01/2007</td>
<td>Active</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
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<td>STC no.</td>
<td>Serial number</td>
<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.IM.A.S.02126</td>
<td>ASSOCIATED AIR CENTER, L.P.</td>
<td>FAA STC ST10613SC</td>
<td>A319-115</td>
<td>Executive style interior (validation of FAA STC ST10613SC)</td>
<td>Index List ref.: 2393000001, dated March 8, 2007 or later approved revision</td>
<td>Master Drawing List, ref.: 2393000030E, dated March 13, 2007 or later approved revision</td>
<td>Instruction for Continued Airworthiness, Chapter 3 &quot;Amendment Limitation&quot;, ref. 2393050000, dated March 8, 2007 or later approved revision</td>
<td>19/03/2007</td>
<td>Active</td>
<td>10015975</td>
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<tr>
<td>EASA.IM.A.S.02128</td>
<td>ELLIOTT AVIATION</td>
<td>FAA STC ST5990CH-T</td>
<td>FALCON 10</td>
<td>ST02333CH: Installation of a Universal Avionics Systems Corporation(UASC) UNS-1K+/L FLIGHT MANAGEMENT SYSTEM (FMS), Dual Universal Avionics Radio Control Units (RCU), and/or dual Rockwell Collins VIR-32/32 Navigation Receivers</td>
<td>- MDL-2651-000 Rev C, dd 19 Oct 2006—AFM supplement(AFMS)/ Instructions for Continued Airworthiness (ICAW) as per MDL-2651-000 Rev C—AFMS FMS-2651-000 Rev IR. dtd 18 Jul 2006 or later approved revision OR—AFMS FMS-2651-001 Rev A. dtd 19 Oct 2006 or later approved revision—ICAW ICA-2651-000 Rev...</td>
<td>1. Certification basis FAR 25 Amdt 117</td>
<td>13/02/2007</td>
<td>Active</td>
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<td>EASA.IM.A.S.02131</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SA0087SE</td>
<td>CESSNA U206G CESSNA U206F CESSNA U206E CESSNA U206D CESSNA U206A CESSNA U206B CESSNA U206 CESSNA TU206F CESSNA TU206D CESSNA TU206C CESSNA TU206B CESSNA TU206A CESSNA TP206E CESSNA TP206D CESSNA TP206C CESSNA TP206B CESSNA TP206A</td>
<td>Installation of vortex generators on the wings, tail surfaces and vertical fin</td>
<td>Cessna STC 7127</td>
<td>In accordance with Micro Aerodynamics Drawing Package MA2146, Revision A, dated March 11, 2004 and installed in accordance with Installation Manual MA2147, Revision IR, dated January 25, 2001 (for 206 series) or MA2196, Revision IR, dated March 11, 2004 (for 207 series), or later approved revisions...</td>
<td>1. If more than Five vortex generators are damaged or missing, the aircraft is not airworthy.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be d...</td>
<td>20/09/2006</td>
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<td>10015979</td>
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<td>STC No</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<tr>
<td>EASA/I.M.A.S.02132</td>
<td>DEL-4IR</td>
<td>CESSNA A152</td>
<td>Installation of Long Range Wing Fuel Tanks</td>
<td>FAA STC</td>
<td>1. If the holder agrees to allow another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>14/04/2006</td>
<td>Active</td>
<td>10115903</td>
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<tr>
<td>EASA/I.M.A.S.02133</td>
<td>HORTON STOL-CRAFT INC.</td>
<td>CESSNA 170B</td>
<td>Installation of STOL Kit(Wing leading edge cuffs, drooped tips, stall fences and aileron gap seals)</td>
<td>FAA STC</td>
<td>1. Aircraft to be placarded &quot;Intentional Spins Prohibited&quot;. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer...</td>
<td>04/09/2006</td>
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<td>EASA/I.M.A.S.02135</td>
<td>FIELD AVIATION COMPANY INC.</td>
<td>BOMBERDIER DHC-6 SERIES 300</td>
<td>Roof Access Steps Installation</td>
<td>FAA STC</td>
<td>--</td>
<td>13/11/2006</td>
<td>Active</td>
<td>10115903</td>
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<tr>
<td>EASA/I.M.A.S.02136</td>
<td>DEL-4IR</td>
<td>CESSNA A152</td>
<td>Installation of a 7 Gallon extended range tank in each wing.</td>
<td>FAA STC</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer...</td>
<td>18/06/2006</td>
<td>Active</td>
<td>10115904</td>
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<td>EASA/I.M.A.S.02138</td>
<td>HARTZELL SERVICE CENTER</td>
<td>PIPER PA-28RT-201T</td>
<td>Installation of a Hartzell PHC-C3YF-1RF/ F7663-4T propeller and C-3570(P) spinner in accordance with Hartzell installation instructions HP2834 dated February 3, 1999 or later approved revision (validation of FAA STC SA1024CH)</td>
<td>FAA approved Airplane Flight Manual Supplement HP383-AFM9 dated February 11, 1999 or later approved revision, is required for PA-28RT-201T (Turbo Arrow II) FAA approved Airplane Flight Manual Supplement HP384-AFM9 dated February 12, 1999 or later approved revision, is required for PA-28RT-201T (Turbo...</td>
<td>08/11/2006</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.02139</td>
<td>O&amp;B AIRCRAFT MODIFICATIONS, INC.</td>
<td>FAA STD-SA1035NE</td>
<td>CESSNA 150</td>
<td>Installation of a 14.6 gallon aft fuselage tank in baggage compartment.</td>
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<td>13/10/2006</td>
<td>Active</td>
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<td>EASA.IM.A.S.02141</td>
<td>DUNCAN AVIATION, INC.</td>
<td>ST02092CH-D</td>
<td>HAWKER 800 XP</td>
<td>Installation of a forward cabin entry door and aft cabin reconfiguration</td>
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<td>20/09/2006</td>
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<td>EASA.IM.A.S.02142</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>ST02046CH-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of the Airshow 4000</td>
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<td>14/12/2006</td>
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<td>EASA.IM.A.S.02144</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>ST02047CH-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation in accordance with Gulfstream Master Drawing List GCS11046000, Revision D, dated August 2, 2005, or later approved revisions.</td>
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<td>14/12/2006</td>
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<td>EASA.IM.A.S.02145</td>
<td>AVIATION FABRICATORS, Inc.</td>
<td>FAA SA147SW</td>
<td>HS 125 SERIES</td>
<td>Installation of one, two, three &amp; four-place side facing seat.</td>
<td></td>
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<td>30/10/2006</td>
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<td>EASA.IM.A.S.02146</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>ST00694CH</td>
<td>DC-9-87 (MD-87) DC-9-83 (MD-83) DC-9-82 (MD-82) DC-9-81 (MD-81)</td>
<td>Installation of Dual Honeywell HT9100 GNSS GPS Navigation System.</td>
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<td>21/05/2007</td>
<td>Active</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>EASA.IM.A.S.02147</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST02694AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Passenger Oxygen System</td>
<td>In accordance with Gulfstream Aerospace Corporation Index List GCS1453800, Revision B, dated August 5, 2003 or later approved revision; FAA approved Flight Manual Supplement GCS1453900, Revision NC, dated August 19, 2003 or later approved rev; FAA approved for the -101 configuration</td>
<td></td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>28/08/2006</td>
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<td>EASA.IM.A.S.02148</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST03270AT-D</td>
<td>GULFSTREAM 200</td>
<td>Installation of a protective shroud around the 60Hz inverter</td>
<td>Index List: CEB2 4930000, Rev. B or later; Instructions for Continued Airworthiness: CEB2 493A000</td>
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<td>21/11/2006</td>
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<td>EASA.IM.A.S.02152</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA09124AC-D</td>
<td>PIPER PA-60-602P, PIPER PA-60-601P</td>
<td>Installation of S-TEC 55/55X two axis automatic flight control system in Aerostar models</td>
<td>- STC SA09124AC-D - AFM Suppl. P/N99124S Rev. 2 - ICA P/N 81319 Rev. 2 - S-TEC Bulletin 709 Rev. 2 - MDL No. 92812 Rev. B</td>
<td></td>
<td>1. According to AFMS, Section 2 Operating Limitations. 2. Instructions for Continued Airworthiness acc. Doc. P/N 81319 Rev. 23. Maintenance Instructions 2-4 must be performed every 100 flight hours or once every calendar year, whichever comes first. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>29/08/2006</td>
<td>Active</td>
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<td>EASA.IM.A.S.02154</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC C-LSA07-003D</td>
<td>BOMBARDIER BD-100-1A10 (CHALLE)</td>
<td>Aircraft completion, 10 passenger interior configuration, in accordance with MDS 070000 (Original STC number: Transport Canada STCC-LSA07-003D)</td>
<td>AFM Supplement F070090.</td>
<td></td>
<td>Limited to S/N 20111.</td>
<td>12/01/2007</td>
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<td>EASA.IM.A.S.02155</td>
<td>TIMCO AVIATION SERVICES INC.</td>
<td>FAA STC ST03342AT-D</td>
<td>BOEING MD-88, BOEING DC-9-87 (MD-87)</td>
<td>Installation of a Door Safety Strap</td>
<td>TIMCO Master data list 06T 055071, Revision B dated October 11, 2006 or later FAA approved revision</td>
<td></td>
<td>1. TIMCO's Instructions for Continued Airworthiness 06T055071 Rev. IR dated June 8, 2006, is a required part of this STC. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>24/10/2006</td>
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<td>EASA.IM.A.S.02157</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA STC SA08-88</td>
<td>CL-600-2B16</td>
<td>Activation of Runway Awareness Advisory System (RAAS) Functionality.</td>
<td>TCCA STC SA08-88, dated 16-11-2006 or later approved revision; Aero Consulting Services Modification Data Summary F036000 Revision NC or later approved revisions; Aero Consulting Services AFM Supplement F036991 Revision NC or later approved revisions.</td>
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<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>TIMCO AVIATION SERVICES INC.</td>
<td>STO12194T-D</td>
<td>AIRBUS A300</td>
<td>Installation of Airbus 210 System</td>
<td>TIMCO Master data list, MSL-01094, Rev LConfiguration Document, CFG-01105, Rev E</td>
<td>Aircraft Serial BB-1547 only/The UNS-1F Flight Management System is approved for En-route, Terminal and Non Precision Approach Operations only</td>
<td>08/10/2006</td>
<td>Active</td>
<td>1016002</td>
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<td>FIELD AVIATION COMPANY Inc.</td>
<td>O-LSA06-175/ D ISSUE 1</td>
<td>BEECH B200 SUPER KING AIR</td>
<td>Installation of a single Universal UNS-1F Flight Management System complete with dual Universal GLS-1250 DOPS Receivers(SCAT-1), Transport Canada LSTC O-LSA06-1769 issue 2.</td>
<td>Configuration AUNS-1F c/w GLS-1250 Electrical and structural Provisions- Installation of the UNS-1F retrofit modifications must be performed in accordance with FADL Master Drawing List designatedDDL-1163 Rev IR, or later approved revisions Installation of the GLS-1250 retrofit modifications...</td>
<td>Aircraft Serial BB-1547 only/The UNS-1F Flight Management System is approved for En-route, Terminal and Non Precision Approach Operations only</td>
<td>05/02/2007</td>
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<tr>
<td>AVIADISEIGN Inc.</td>
<td>FAA STC SA0783NM</td>
<td>BEECH B200IT, 300, 1900, 1900C, BEECH B200, B200C, B200T, BEECH 200, 200C, 200CT, 2001T, 2001C, 2001T</td>
<td>Installation of upper wing reinforcing kit.</td>
<td>In accordance with FAA sealted Aviadesign Inc. Master Drawing List E.O.-88580, dated March 10, 1988, or later approved revision.</td>
<td>1. Reinforcing strap and original wing structure covered by the new strip tube inspected every 1000 flying hours or annually for any signs of cracking or corrosion 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;...</td>
<td>19/09/2006</td>
<td>Active</td>
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<td>AVIATION FABRICATORS, Inc.</td>
<td>FAA STC ST01139W</td>
<td>560</td>
<td>Installation of a 39-inch or a 45-inch two-place divan or 42-inch one-place divan per Aviation Fabricators STC Data Package, AF-342, Rev A, dated 12, June, 2003 and stamped &quot;FAA Approved&quot; with an approved date of October 22, 2003 or later approved revision</td>
<td>STC Data Package, AF-342, Rev A, dated 12, June, 2003 and stamped &quot;FAA Approved&quot; with an approved date of October 22, 2003 or later approved revision</td>
<td>The STC covers only the installation of parts contained in the Description section/phototoc, covers, aisle width, oxygen, seat belts are not covered</td>
<td>13/09/2006</td>
<td>Active</td>
<td>1016005</td>
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<td>CLIPPER AVIATION, Inc.</td>
<td>FAA STC ST02318NY</td>
<td>BOEING 747-400 FREIGHTER</td>
<td>Upper Deck seating arrangement for B747-400BCF as defined in Weser Aircraft installation drawing No. 960014 and ClipperAviation CLI-2006-003 Rev. A, Master Data List B747-400BCF</td>
<td>CL-52006-007 Airplane Flight Manual Supplement for Boeing 747-400 Special Freighter</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>04/04/2007</td>
<td>Active</td>
<td>1016006</td>
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<td>VISIONSAFE CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST0105W1</td>
<td>BD-700-1A10, BD-700-1A11</td>
<td>Installation of Emergency Vision Assurance System (EVAS) Model: 107STC-191 and 107 STC-192</td>
<td>-MDL: Midcoast Aviation 35-4167201 Rev A or Vision Safe No. 20 Rev H- AFM-Supplement: Midcoast Aviation 35-4167202 Rev Originl or VisionSafe No. 1. Rev 1 or later revisions of theabove listed documents approved by EASA inaccordance with EASA ED Decision 2004/04/05 (or subsequent revisions offthis de...</td>
<td>10/02/2009</td>
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<td>EASA.IM.A.S.02173</td>
<td>OGN AEROSPACE CHEM-TRONICS Inc.</td>
<td>FAA STC ST01168A</td>
<td>B &amp; E; 9-12-2006</td>
<td>BOEING MD-88</td>
<td>Installation of ASTECH Manufacturing, Inc. inlet assembly</td>
<td>STC SA2684CE</td>
<td>DHC-8-1FMS at Revision 1 or later TCCA approved revision. Operation must be in accordance with Spectrum Aeronautical Inc. Master Document List 6039-1V-A revision 1 or later TCCA approved revision. Maintenance must be in accordance with TCCA Limited STC C-LSA06-335/D to S/N 992-1953-001 Rev. C.</td>
<td>31/10/2006</td>
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<td>10016010</td>
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<tr>
<td>EASA.IM.A.S.02175</td>
<td>JCM AERODESIGN LIMITED</td>
<td>O-LSA08-268/D ISSUE 1</td>
<td>BOMBARDIER DHC-8-311</td>
<td>EMS Installation as per TCCA Limited Supplemental Type Certificate O-LSA08 268/D, Issue 1, dated 13 September 2006</td>
<td>Installation must be in accordance with Spectrum Aeronautical Inc. Master Document List 6039-1V-A revision 1 or later TCCA approved revision. Operation must be in accordance with Spectrum Aeronautical Inc. Flight Manual Supplement 6039-1FMS at Revision 1 or later TCCA approved revision. Maintenance must be in accordance with TCCA Limited STC C-LSA06-335/D to S/N 992-1953-001 Rev. C.</td>
<td>20/12/2006</td>
<td>Active</td>
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<td>EASA.IM.A.S.02176</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Aircraft Completion, 13 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA08-335/D to S/N 992-1953-001 Rev. C.</td>
<td>Approved Model List No. CMR-750-115, of 20 June 2006. Installation of a Honeywell DL-950 Data Loaderto replace the existing DL-800.</td>
<td>Refer to Limitations sections of the required FM Supplement. The use of WLAN is subject to operational rules.</td>
<td>05/12/2006</td>
<td>Active</td>
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<tr>
<td>EASA.IM.A.S.02179</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC SA2683CE</td>
<td>BOEING MD-88</td>
<td>BOEING DC-9-87 (MD-87) BOEING DC-8-83 (MD-83) BOEING DC-8-92 (MD-82) BOEING DC-8-81 (MD-81)</td>
<td>Installation of Collin Traffic Alert and Collision Avoidance System (TCAS I)</td>
<td>STC No. SA2683CE Drawing Index DWG No. 992-1592-001 (Rev. J) FAA STC SA2683CE</td>
<td>Installation of Mode S Transponder System STC EASA.IM.A.S.03810 (SA2683CE) or EASA. IM.A.S. 02181 (SA3082SO) is a prerequisite to this modification.</td>
<td>04/01/2007</td>
<td>Active</td>
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<tr>
<td>EASA.IM.A.S.02180</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC SA3082SO</td>
<td>BOEING MD-88</td>
<td>BOEING DC-9-83 (MD-83) BOEING DC-8-92 (MD-82) BOEING DC-8-91 (MD-81)</td>
<td>Installation of Dual Collins TPR-720 Mode S Transponders</td>
<td>STC SA3082SO Master Drawing List MDL-91-47 Rev. C.</td>
<td>None. Registration (if applicable) - Serial number (if applicable)</td>
<td>04/01/2007</td>
<td>Active</td>
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<td>EASA.IM.A.S.02181</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC SA2684CE</td>
<td>DC-9-81 (MD-81)</td>
<td>DC-9-83 (MD-83) BOEING DC-9-92 (MD-82) BOEING DC-8-81 (MD-81)</td>
<td>Installation of Dual Collins Mode S Transponders</td>
<td>STC SA2684CE Drawing Index DWG No. 982-1953-001 Rev. C.</td>
<td>None. Registration (if applicable) - Serial number (if applicable)</td>
<td>04/01/2007</td>
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<td>EASA.IM.A.S.02182</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>ST014615C-D</td>
<td>FALCON 900</td>
<td>Mid Cabin Door Installation, Envite</td>
<td>STC SA2685CE Drawing Index DWG No. 992-1592-001 (Rev. J)</td>
<td>This STC is Limited to F9EX-01056-501 installations in Model Falcon 900EX.</td>
<td>27/10/2006</td>
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<td>EASA.IM.A.S.02184</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>ST102530C-D</td>
<td>FALCON 900EX</td>
<td>Oxygen System Installation.</td>
<td>STC SA2684CE Drawing Index DWG No. 992-1592-001 (Rev. J)</td>
<td>This STC is Limited to F9EX-01056-501 installations in Model Falcon 900EX.</td>
<td>27/10/2006</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<tr>
<td>EASA.IM.A.S.02157</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>REV. 1</td>
<td>FAA STC ST03961AT-D</td>
<td>GIV</td>
<td>Installation of an Enhanced Surveillance Flight Identification Systems w/ Collins Radio Tuning Unit.</td>
<td>Rev. 1: Typo connection.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>15/03/2007</td>
<td>Active</td>
<td>1011623</td>
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<tr>
<td>EASA.IM.A.S.02158</td>
<td>SHADIN AVIONICS</td>
<td>ST02115CH</td>
<td>CESSNA 550</td>
<td>GULFSTREAM GIV</td>
<td>Validation of FAA STC ST02115CH Installation of a dual Shadin &quot;ADC 600 System RVSM Capable&quot; hardware provisions for RVSM capability.</td>
<td>Master Data List D2M-076-002, Rev. H. 15 October 2005 or later approved revision and Report 2807A-AFM Supplement: see &quot;Limitations and Conditions&quot;.</td>
<td>This STC does not provide RVSM capability to the aircraft (compliance to RVSM airworthiness requirements are not part of this STC) a separate STC has to be applied. Airplane Flight Manual Supplement, Shadin Document No. D2M-076-002, Revision B, dated 10 November 2005 or later approved revision is required.</td>
<td>07/12/2006</td>
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<td>EASA.IM.A.S.02191</td>
<td>TELAR INTERNATIONAL, LLC</td>
<td>REV. 1</td>
<td>FAA STC ST06587 LA</td>
<td>BOEING 747-400F BOEING 747-400</td>
<td>Installation of a Cargo Handling System on the lower deck of the B747-400F, 400F.</td>
<td>FAA approved Telaire International Index List IL 195108 Rev. C dated May 24, 1999 or later FAA approved revisions.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is not permitted.</td>
<td>12/01/2009</td>
<td>Active</td>
<td>1011623</td>
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<tr>
<td>EASA.IM.A.S.02192</td>
<td>BEECHCRAFT CORPORATION</td>
<td>REV. 2</td>
<td>FAA ST104326C-D</td>
<td>HAWKER 800 XP HAWKER 800 XP</td>
<td>Installation of Mark V Enhance Ground Proximity Warning System (Egwps) with and without Runway Awareness Advisory System.</td>
<td>1. In accordance with Master Drawing List 800-514 Revision D or later approved revision. 2. Flight Manual Supplement 800-529 Revision R6 or later approved revision. 3. Maintenance Manual Addenda 34-40-40.</td>
<td>1. The Flight Manual Supplement is not needed for the -101 installation because the aircraft's basic AFM is still applicable. 2. When fitted to Hawker 800XP and 800XP this STC is limited to the following aircraft: 258541, 258556, 258667 through 258690, 258611 through 258628, 258630 through 258688.</td>
<td>21/06/2008</td>
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<td>Limitations/Conditions</td>
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<td>LANDMARK AVIATION</td>
<td>ST00938CH-D</td>
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<td>FALCON 10</td>
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<td>Installation of a Honeywell Enhanced Ground Proximity Warning System (EGPWS) in accordance with Garrett Aviation Drawing List DL309 Revision F, dated 20th January 2005.</td>
<td>FAA STC ST00938CH-D, Airplane Flight Manual Supplement AFMS3201 Rev B for configurations 1,2 and 3</td>
<td>FAA STC ST00938CH-D, Airplane Flight Manual Supplement 100263B Rev B for configurations 4 and 6</td>
<td>Certification Considerations for the Terrain Awareness Warning System: TAWS requirements have been applied. Configurations 2 and 3 (without Terrain Display) do not qualify for a Class A TAWS equip...</td>
<td>06/02/2007</td>
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<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>TCCA STC</td>
<td>SA01-30</td>
<td>DHC-6 100, DHC-6 200, DHC-6 300</td>
<td></td>
<td>Removal of Auxiliary Battery</td>
<td>Compliance Checklist CB2796.</td>
<td>1. An approved Main Battery Installation with a minimum capacity of 34 Ah. Hours must be embodied as a pre-requisite. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft...</td>
<td>22/12/2006</td>
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<tr>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>TCCA STC</td>
<td>SA01-17</td>
<td>BOMBARDIER DHC-6 SERIES 300</td>
<td></td>
<td>Cabin Evacuation Lighting Control System Change</td>
<td>Transport Canada STC SA01-17 Issue 349r1 Data List MSL:1289-01Avro Design Ltd Flight Manual Supplement FM3396.02</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>10/10/2008</td>
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<td>TELAR INTERNATIONAL, LLC</td>
<td>REV. 1</td>
<td>FAA STC</td>
<td>BOEING 747-400SF</td>
<td>ST0456LA</td>
<td>Installation of a Cargo Handling System on the main deck of the B747-400SF in accordance with FAA approved Telar International Index List ER 19/100 Rev. N/Cdated October 21, 1997 or later approved revisions. Validation of FAA STC ST0456LA</td>
<td>FAA STC ST0456LA, FAA Approved Airplane Flight Manual Supplement 100263B Rev B for configurations 4 and 6</td>
<td>FAA STC ST0456LA, FAA Approved Airplane Flight Manual Supplement 100263B Rev B for configurations 4 and 6</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce...</td>
<td>11/12/2008</td>
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<tr>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>FAA STC</td>
<td>SA02297NY</td>
<td>DE HAVILLAND CANADA</td>
<td>DHC-300</td>
<td>Installation of a Universal Avionics CVR-30B Cockpit Voice Recorder. (1)</td>
<td>Installation of a Honeywell SSCVR, or L3 FA2100 Cockpit Voice Recorder.</td>
<td>FAA STC SA02297NY, FAA Approved Airplane Flight Manual Supplement 100263B Rev B for configurations 4 and 6</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>22/10/2007</td>
<td>Active</td>
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<p>| EASA:IMA:S.02204 | ROSEN SURVIVOR SYSTEMS LLC | FAA STC SA1478M | HAWKER 900 XP | Installation of Cockpit Sun Visors. | In accordance with Rosen drawing List Number RBA-600-DL, Revision B, dated May 10, 2007, or later approved revisions. | The installer must determine, if this design is compatible with previously approved modifications. If the holder agrees to permit another person to use this certification or altera product, the holder must give the other person written evidence of that permission. | 21/06/2008 | Active | 10/11035 |
| EASA:IMA:S.02206 | LANDMARK AVIATION | FAA STC ST01579C-D | MYSTERE FALCON 20- E5 | Installation of a Honeywell MK-VIII Enhanced Ground Proximity Warning System (EGPWS) in accordance with Garrett Aviation Services Drawing List 70-0177-001 revision G dated 20th October 2005 or later approved revisions. | FAA STC ST01579C-D - Airplane Flight Manual Supplement 76-8739-003 Revision B or later approved revisions for aircraft configuration 01... | 03/07/2007 | Active | 10/11037 |
| | | FAA STC ST01705E | BOEING 747-400 | Removal of the strengthened flight deck door and modification and reinstallation of baggage bins on the upper deck in accordance with JACO Aero Design &amp; Engineering Master Drawing List “JD-03A02 Rev. F” or later revisions of the above listed documents approved by EASA in accordance with EASA ED de... | 1. JD-03A02 Rev. F Master Drawing List B747-450 Pax to Freight upper deck reconfiguration. 2. JD-03A02 Rev. F Master Drawing List B747-25-V554 Rev. F JADE Service Bulletin Pax to Freight upper deck reconfiguration. 3. FAA STC ST01705E - Airplane Flight Manual Supplement - | 1. This STC is limited to the following aircraft after incorporation of Boeing SB 747-00-2004 Passenger to Freighter conversion a. RT561 Sn. 24061b. RT566 Sn. 24066c. RT507 Sn. 24265d. RT508 Sn. 242752. This STC is approved only for the product configuration. | 02/04/2007 | Active | 10/11040 |
| EASA:IMA:S.02210 | WESTERN AVIONICS | FAA STC SA01555 ISSUE 5 | DHCS-300 | Installation or Upgrade of a Honeywell MK VI Enhanced Ground Proximity Warning System (EGPWS) | For installation: * Transport Canada approved Western Avionics Engineering Order No. 91429-401, Revision 2, dated 16-Mar-2006, or later approved issue/or Upgrade. * Transport Canada approved Western Avionics Engineering Order No. 91429-402, Revision 2, dated 16-Mar-2006, or later app... | | 12/01/2007 | Active | 10/11041 |</p>
<table>
<thead>
<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>Rev. level</th>
<th>Foreign/NAA STC ref.</th>
<th>Model(s)</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tbody>
<tr>
<td>EASA.IM.A.S.02213</td>
<td>HONEYWELL INTERNATIONAL Inc.</td>
<td>FAA STC ST01003CH-D</td>
<td>F3B, F3B/RA, 400A, 400B, 403B, DH-125 SERIES 1A, 3A, 3ARA, 600A, 600B, 700A, 700B</td>
<td>Installation of Honeywell N1 Digital Electronic Engine Control (DEEC).</td>
<td>Installation of Honeywell N1 Digital Electronic Engine Control (DEEC) in accordance with Garrett Aviation Services Drawing List DL1003, dated 26 Aug. 1999, no revision, or later approved revisions. Operation to be in accordance with an approved Airplane Flight Manual Supplement, document AFM303 date...</td>
<td>None</td>
<td>Compatibility of this modification with other previously approved modifications must be determined by the installer. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft...</td>
<td>12/12/2006</td>
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<td>EASA.IM.A.S.02214</td>
<td>DC AEROSPACE</td>
<td>REV. 1</td>
<td>ST03011AT</td>
<td>BOEING 727-200 SERIES / BOEING 727-100 SERIES</td>
<td>Installation of Sandel ST3400 Terrain Awareness and Warning System (TAWS) in Boeing 727 Series Aircraft according to FAA STC ST03011AT dated 22 December 2004</td>
<td>- MDL-2418, Revision H dated 25 July 2005 - AFM Supplement FMS-2418-1, Rev. 9 dated 23 December 2004 - Instructions for Continued Airworthiness ICA-2418, Rev. B dated 17 December 2004 or later revisions of the above listed documents approved by EASA in accordance with EASA ED decision 2004/044CF (or a...</td>
<td>SANDEL mandatory SB 3400-09 to be incorporated. The limitations as listed in FAA STC ST03011AT apply.</td>
<td>23/07/2007</td>
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<td>EASA.IM.A.S.02215</td>
<td>DASSAULT FALCON JET CORP (SERVICE CENTER)</td>
<td>ST09737SC-D</td>
<td>FALCON 900EX</td>
<td>Bracket Installation for Mid Cabin Door i.e.w. Drawing List Report “F90-01093”</td>
<td>None</td>
<td>None</td>
<td>Vision Safe Corporation Instructions for Continued Airworthiness, Rev. 04, dated March 2006.</td>
<td>SANDEL mandatory SB 3400-09 to be incorporated. The limitations as listed in FAA STC ST03011AT apply.</td>
<td>19/12/2006</td>
<td>Active</td>
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<td>EASA.IM.A.S.02217</td>
<td>HAWKER BEECHCRAFT SERVICES, Inc.</td>
<td>FAA STC SA04785C</td>
<td>BEECH B300, 8300C / BEECH B300GT, B200CGT / BEECH B200, B200C</td>
<td>Installation of Digital Flight Data Recorder System</td>
<td>Master Data List R050050131 Revision 12, 16 April, 2009 (w/o approved revision) and FAA approved revision. Operation to be in accordance with an approved Airplane Flight Manual Supplement R050050132 Revision 2.0, dated 26 March, 2009 (w/o approved revision).</td>
<td>None</td>
<td>None</td>
<td>Restricted to ASN 5636.</td>
<td>16/06/2009</td>
<td>Active</td>
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<td>EASA.IM.A.S.02218</td>
<td>BOMBARDIER AEROSPACE</td>
<td>REV. 1</td>
<td>TCCA SA03-113</td>
<td>CL-600-2B16</td>
<td>Complete Custom Interior</td>
<td>Transport Canada STC SA03-113. Aer Consulting Services Configuration Definition List DA0700 Revision W dated November 13, 2006. Aer Consulting Services AFM Supplement DA07000 Revision N/C or later approved revision.</td>
<td>Restricted to ASN 5636.</td>
<td>17/01/2007</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.02220</td>
<td>LIVETV, L.L.C.</td>
<td>FAA STC ST10056SC</td>
<td>AIRBUS A320-232</td>
<td>Installation Of Electronic Equipment Rack</td>
<td>Master Data List, Document Number 2284-001 Revision N, dated 03rd March 2003 or later EASA approved revision. Instructions for Continued Airworthiness, Document Number CD-2002-002, Revision R, dated 14th August 2002, or later EASA approved revision. Aircraft Flight Manual Supplement, Document Number...</td>
<td>As per FAA STC ST10065SC the following STC's must be installed concurrently with this STC: ST00788SE, ST02483AT and ST01447NY.</td>
<td>29/05/2007</td>
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<td>EASA.IM.A.S.02221</td>
<td>LIVETV, L.L.C.</td>
<td>FAA STC ST02483AT</td>
<td>AIRBUS A320 SERIES</td>
<td>Installation Of a Cabin Mounted Video Camera System</td>
<td>Master Drawing List, Document Number LYTU-002-MDL, Revision D, dated 07th April 2003, or later EASA approved revision. Instructions for Continued Airworthiness, Document Number LYTU-002-CAW, Revision B, dated 22nd February 2002, or later EASA approved revision. Aircraft Flight Manual Supplement, Document...</td>
<td>As per FAA STC ST10065SC the following STC's must be installed concurrently with this STC: ST00788SE, ST02483AT and ST01447NY.</td>
<td>29/05/2007</td>
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<td>EASA.IM.A.S.02222</td>
<td>LIVETV, L.L.C.</td>
<td>FAA STC ST00788SE</td>
<td>AIRBUS A320-200</td>
<td>Installation of a Live TV Satellite Television System</td>
<td>Master Drawing List, Document Number LTV933-D01-A320, Revision J, dated 24th May 2005 or later EASA approved revision.</td>
<td>As per FAA STC ST10065SC the following STC's must be installed concurrently with this STC: ST00788SE, ST02483AT and ST01447NY.</td>
<td>29/05/2007</td>
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<td>EASA.IM.A.S.02223</td>
<td>LIVETV, L.L.C.</td>
<td>FAA STC ST01447NY</td>
<td>AIRBUS A320-200</td>
<td>Installation Of Harris Ground Wireless Aircraft Data Link &amp; Partial Power Provisions for Installation of a Cabin Surveillance System</td>
<td>Master Drawing List, Document Number 3092086-MDL, Revision C, dated 29th July 2004, or later EASA approved revision. Instructions for Continued Airworthiness, Document Number 3092085, Revision B, dated 31st January 2002, or later EASA approved revision.</td>
<td>As per FAA STC ST10065SC the following STC's must be installed concurrently with this STC: ST00788SE, ST02483AT and ST01447NY.</td>
<td>29/05/2007</td>
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<td>10016051</td>
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<td>EASA.IM.A.S.02224</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA09218AC-D</td>
<td>CESSNA 182T CESSNA 182R CESSNA 182Q</td>
<td>Installation of S-TEC System 20/30 Single and Two Axis Automatic Flight Guidance System, Model ST-718-20/30 in Cessna 182Q, 182R and T182, also when modified per STC SA1382WE (Robertson STOL)</td>
<td>Bulletin No. 818, dated 30- Jul-1997 or later approved revision. Master Drawing List No. 921016, dated 06- Aug-1997, or later approved revision.</td>
<td>For 182Q the maximum aft c.g. is limited to 46.0 inches aft of datum when autopilot is installed. Installer must check availability of EASA-approval for Robertson STOL STC No. when airplane is modified per that STC.</td>
<td>14/12/2006</td>
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<td>Associated Technical Documentation</td>
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<td>EASA.IM.S.02229</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA0916AC-D</td>
<td>FAAC STC</td>
<td>CESSNA TR182</td>
<td>CESSNA R182</td>
<td>Installation of S-TEC System 2030 Single and Two Axis Automatic Flight Guidance System, Model ST-759-23/30 in Cessna R182, and TR182.</td>
<td>- FAA approved RAS Master Drawing List No. 9210S9, dated 15-Dec-1997 or later approved revision. - Master Drawing List No. 9210S9, dated 15-Dec-1997, or later approved revision. - FAA/ DAS Approved Pilot's Operating Handbook and / or Airplane Flight Manual Supplemental, P/N 891615, dated 16-Dec-1997, or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>18/12/2006</td>
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<td>EASA.IM.S.02227</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC C-LSA07-086D REV NC</td>
<td>BOMBARDIER</td>
<td>BOMBERDIER</td>
<td>BD-700-1A11</td>
<td>Customized Interior Completion - SN 9209/validiation of TCCA STC C-LSA07-086D Rev NC</td>
<td>APM Supplement G-FM2500015TC4A Stc Rev C-OS-06-514, dated June 12, 2007CA. G007300000-FCA</td>
<td>1. Refer to the Limitations sections of the required FM Supplement.2. WLAN system is compatible only to IEEE 802.11(b).3. The use of WLAN is subject to operational rules.</td>
<td>21/06/2007</td>
<td>Active</td>
<td>10102055</td>
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<td>EASA.IM.S.02233</td>
<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA STC SA1404WI</td>
<td>BEECH 300, B300C</td>
<td>BEECH 200, 200C, B200C</td>
<td>Installation of Collins IRIS 5000</td>
<td>FAA approved RAS Master Data List, Doc. No. RAS05060001, Revision 5.0, dated 30-Aug-2006 or later approved revision. FAA approved Airplane Flight Manual Supplement, Doc. No. RAS05060034, dated 19-Sep-2006, or later approved revision.</td>
<td>This change is limited to airplanes equipped with Collins Pro Line 21 System and includes the FSU-5010 File Server Unit, CCP-3000 Cursor Control panel and either the XMVR 1000 XM Weather Receiver or VHF 4000/ CMU4000 Universal Weather System and associated antenna.</td>
<td>15/01/2007</td>
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<td>EASA.IM.S.02244</td>
<td>AS&amp;T AEROSPACE SYSTEM &amp; TECHNOLOGIEIN C</td>
<td>FAA SA756GL</td>
<td>U206F, U206G</td>
<td>U206F, T206H, T206G</td>
<td>Installation of TKS Ice Protection Systems Components on a no-hazard / no credit basis.</td>
<td>AS&amp;T Drawing List No. 02800-DL, Revision 3, dated 19 December 2003, or later approved revision. &quot;Supplemental Airplane Flight Manual and FAA approved Airplane Flight Manual Supplement&quot;, Doc. No. 02800-FMS, revised May 27, 2003 or later approved revision.</td>
<td>1. This installation is not approved for flight into known icing conditions.2. A placard must be fitted on the upper control panel in front of the pilot, stating &quot;FLIGHT INTO KNOWN ICING CONDITIONS IS PROHIBITED&quot;.3. This installation is approved on a no-hazard / no credit basis.4. This STC is approved for engine configurations shall be determined by the installer.</td>
<td>04/01/2007</td>
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<td>10102057</td>
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<td>EASA.IM.S.02236</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>ST01580AT-D</td>
<td>BOEING 747-460</td>
<td>Installation of radio panels and Collins VHF transceivers with 8.33 kHz spacing in accordance with Collins Master Data List MDL-01237, revision M, dated August 9, 2005, or later FAA approved revisions.</td>
<td>Data package containing the following documents, and additional supporting data: MDL 01237, rev. M. MDL Collins VHF700V-900V in B747-400; CFG-01296; installation of Collins VHF-700B communications system on B747-400 for Air France ENG-01664; installation Collins VHF-700B on Boeing 747-400.</td>
<td>1. This installation is approved for engine configurations shall be determined by the installer.</td>
<td>08/08/2007</td>
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<td>EASA.IM.A.S.02237</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST04100CH-D</td>
<td>BAE 125 SERIES 800A</td>
<td>Installation of Rockwell Collins Pro Line 21 IDS (Integrated Display System) with IFIS-5000 (Integrated Flight Information System) and Engine Instrument Display</td>
<td>Installation in accordance with Duncan Aviation Master Drawing List 060912014, Revision G, dated 4/20/07, or later approved revisions. Aircraft to be operated in accordance with Aircraft Flight Manual Supplement, 060912018, Revision B dated 1/18/07, or later approved revisions. Aircraft to be operated ...</td>
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<td>28/02/2008</td>
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<td>EASA.IM.A.S.02239</td>
<td>DUGAN AIR TECHNOLOGIES, INC</td>
<td>FAA STC ST01825E</td>
<td>737-500, 737-200, 737-300, 737-400</td>
<td>Installation of Duganair Security Flight Deck Door as defined in FAA STC ST01825E</td>
<td>- Duganair Technologies Inc., Index List, QS-0001-3A, Revision B, dated August 27, 2003 or later approved revisions.- FAA Approved Airplane Flight Manual Supplement No. QWC737101-4 Rev. 1, dated December 16, 2004 or later approved revision is required as part of this installation. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
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<td>15/11/2006</td>
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<td>EASA.IM.A.S.02240</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCAC LISTC CL-600-2B16</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Costumised Interior Completion</td>
<td>Transport Canada LSTC C-LSA07-044/Aero Consulting ServicesModification Data Summary E964000 Revision m/c or later approved revision. FAA Approved Aero Consulting Services AFM Supplement E9640920 Revision m/c or later approved revision.</td>
<td>Restricted to ASN 5659.</td>
<td>11/04/2007</td>
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<td>EASA.IM.A.S.02241</td>
<td>NORTHEAST ENGINEERING &amp; DEVELOPMENT LTD</td>
<td>REV. 1</td>
<td>CESSNA T207A</td>
<td>STOL Installation</td>
<td>- Installation must be in accordance with Northeast Engineering &amp; Development Ltd Master Drawing Li...</td>
<td>1. This Installation is limited to the Serial No. 20700706 (Registration:C-FYBZ). 2. Compatibility of this modification with other previously approved modifications must be determined with the installer. 3. This STC is approved only for the product configuration as defined in the approved design data ref...</td>
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<td>EASA.IM.A.S.02242</td>
<td>ANCRA INTERNATIONAL LLC</td>
<td>FAA STC ST01551LA</td>
<td>ATR 42-300 AEROSPATIALE ATR 42-300</td>
<td>Installation of Ancra mechanical main deck cargo landing system</td>
<td>- Ancra Controlled Drawing List DL84230, Revision N/ C, dated June 20, 2003</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data ref...</td>
<td>08/12/2006</td>
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<td>EASA.IM.A.S.02244</td>
<td>AEROMECH Inc.</td>
<td>FAA ST015202E</td>
<td>501, 551</td>
<td>Validation of FAA STC ST015202E Group aircraft approval for RVM/GSM capability.</td>
<td>AeroMech Document No. AM-20-03-0311, Rev. B, dated 08 Sept. 2006 or later FAA approved revision. FAA approved AeroMech AFM-S Doc. AM-AFM-S-CS-0706, dated 04/04/2007 or later FAA approved revision. FAA approved AeroMech Installation for ComKw Doc. No. AMH/02A-C-0170, dated 04/04/2007 or later FAA a...</td>
<td>The installer must determine if this design is compatible with previously approved modifications. If the holder agrees to permit another person to use this certification or alter the product, the holder must give the other person written evidence of that permission.</td>
<td>26/04/2007</td>
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<td>LEARJET Inc.</td>
<td>FAA STC ST01731LA-D</td>
<td>LEARJET 45</td>
<td>Installation of Honeywell Integrated Communication Unit / Enhanced Surveillance Upgrade (-1 Configuration)</td>
<td>Project Specification Certification Plan BA54/337001, rev. NC/JAR/Compliance Checklist BA54/160014, rev. NCMaster Document List BA54/9510/13, rev. EAFM Supplement BA54/9540/08, rev. NCMandatory Instructions for Continued Airworthiness BA54/9510/03-PCA-1, rev. NC</td>
<td>1. Incorporation of ECN 7568 and Honeywell Service Bulletins 7517402-23-6008, 751070-23-6044 and 751070-23-6046 is a prior or concurrent requirement to this installation. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs...</td>
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<td>EASA.IM.A.S.02246</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03324AT-D</td>
<td>GULFSTREAM GIV-SP</td>
<td>Completion and installation of the custom interior on Aircraft Serial 5127.</td>
<td>In accordance with Gulfstream Aerospace Corporation Master Index List GC31-4915127. Revision E, dated January 22, 2007, or later approved revisions. FAA approved Flight Manual Supplement, GC51-501980, dated January 26, 2007, or later approved revisions</td>
<td>Aircraft Serial 5127 Only. The following STCs are required to be installed as part of this STC: FAA STC ST02696AT-D, EASA Approved ref EASA.IM.A.S. 02147. FAA STC ST03098AT-D, EASA Approved ref EASA.IM.A.S.02259FAA ST02697AT-D was approved prior to EASA under the JAA System and are listed in the Gulfs...</td>
<td>05/02/2007</td>
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<td>EASA.IM.A.S.02247</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA LIMITED STC C-LSA06-281 D</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Aircraft Completion, 13 Passenger Interior Configuration/Validation of TCCA Limited STC C-LSA06-281D to S/N 9186/1989, E91500/Rev A or later approved revisions</td>
<td>SuC: C-06-074787MS G-FM2590014 Rev NC or later approved revisions: F/C 9186099000 ICCA I</td>
<td>Refer to the Limitations of Sections of the required PM Supplement/WLAN system is compatible only to IEEE 802.11 (b) The use of WLAN is subject to operational rules.</td>
<td>23/01/2007</td>
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<td>EASA.IM.A.S.02249</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC ST03216CH</td>
<td>DC-9-80 SERIES</td>
<td>FAA STC ST03216CH - Elementary and Enhanced Mode S.</td>
<td>Limited to configuration 3 of ECS MDL ECS-203380 Rev G dated July 7, 2006. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft engine com...</td>
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<td>EASA.IM.A.S.02251</td>
<td>LEARJET Inc.</td>
<td>FAA STC SA81789NM-D</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Installation of an Artex ELT/NAV Interface System/Validation of the FAA STC SA81789NM-D</td>
<td>IFCA: GC-2561000-1 IFCA-1 or -2</td>
<td>21/02/2007</td>
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<td>EASA.IM.A.S.02256</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03351AT-D</td>
<td>GULFSTREAM GN-X</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC41-491060, Revision F, dated 05 January 2007 or later approved revision.</td>
<td>Aircraft Serial Number 4806 Only. The following STCs are a required part of this STC: ST03097AT-D, ST03098AT-D, ST03099AT-D, ST03094AT-D, ST03095AT-D and ST03097AT-D. It is the Installer's responsibility to ensure these STCs are all EASA approved.</td>
<td>23/01/2007</td>
<td>Active</td>
<td>1016774</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
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<tr>
<td>EASA.IM.A.S.02258</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCA LIMITED STC Co.</td>
<td>LSA07-045D</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Customized Interior completion - Aircraft 5664.</td>
<td>Transport Canada Limited STC No. CL-SA07-045D dated 9 June 2007 Aero Consulting Services Modification Data Summary FOA1000 version NC or later approved revision. Aero Consulting Services Airplane Flight Manual Supplement FOA1000 version NC or later approved revision.</td>
<td>Restricted to Aircraft Serial No. 5664.</td>
<td>18/01/2007</td>
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<td>EASA.IM.A.S.02261</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST01901LA</td>
<td></td>
<td>BOEING 767-200I-300</td>
<td>Installation of ARINC 718A MODE S Transponder Activation of Elementary and Enhanced Surveillance Capabilities.</td>
<td>Hollingshead International Master Data List 2869-01, Revision B, No. 13/10/2006 AFM Supplement, Hollingshead International Document 2869-AMF, Revision NC, 17/10/2006.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification must be done by the installer.</td>
<td>31/07/2009</td>
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<td>EASA.IM.A.S.02261</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST01901LA</td>
<td></td>
<td>BOEING 767-300</td>
<td>Installation of ARINC 718A MODE S Transponder Activation of Elementary and Enhanced Surveillance Capabilities.</td>
<td>Hollingshead International Master Data List 2869-01, Revision B, No. 13/10/2006 AFM Supplement, Hollingshead International Document 2869-AMF, Revision NC, 17/10/2006.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/06/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.02262</td>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA SA01567SC</td>
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<td>F90</td>
<td>Installation of Pratt and Whitney PT6A-135A engines in place of PT6A-135 engines.</td>
<td>Installation must be in accordance with Master Drawing List No. 19015-00-00 Rev, D or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement Document No. AFM19015 Rev. D dated 15-10-09 or later approved revision.</td>
<td>1. Installation limited to Serial Number LA-2 thru LA-204, except for LA-202. 2. Compatibility of this modification with other previously approved modifications must be determined by the installer. 3. This STC is approved only for the product configuration as defined in the approved design data...</td>
<td>23/01/2007</td>
<td>Active</td>
<td>1011676</td>
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<td>EASA.IM.A.S.02263</td>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA SA00433AT</td>
<td></td>
<td>BEECH 2007 200, 200C, 200CT, 200T</td>
<td>Installation of Pratt and Whitney PT6A-42 engines in place of PT6A-41 engines.</td>
<td>Installation of Pratt &amp; Whitney PT6A-42 engines with Propellers model HC-B37N-3(G,N,T10178)(B,NB,K,N), or Harvard HC-Delhi-JA66383K installed in accordance with modifications drawing No. A794-001, Rev. B, dated 17 July 2007 or later approved revision. Operation must be in accordance with Airworthiness must be in accordance with FAA STC ST02267CH</td>
<td>01/04/2008</td>
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<td>EASA.IM.A.S.02264</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA90201AC-D</td>
<td>CESSNA 172S</td>
<td>Installation of S-TEC System SAS/550 Two Axis Automatic Flight Guidance System, Model ST-563 in Cessna 172R and 172S</td>
<td>FAA approved revision Master Drawing List No. 92732, Revision E, dated 01-Feb-2001, and later approved revision FAA/DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N 891336, Revision 2, dated 01-Feb-2001, or later...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>16/12/2007</td>
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<td>EASA.IM.A.S.02265</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA C-LSA07-111/D</td>
<td>BOEING 747-400 SERIES</td>
<td>AIRCRAFT COMPLETION, 9 PASSENGER INTERIOR CONFIGURATION/EASA/Validation of Transport Canada Limited Supplemental Type Certificate C-LSA07-111/D Issue 1 dated 11th May 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F111050 Revision N/C.</td>
<td>Transport Canada Limited Supplemental Type Certificate C-LSA07-111/D Issue 1 dated 11th May 2007, Transport Canada Approved Aero Consulting Services Ltd. Airplane Flight Manual Supplement F111050 Revision N/C. Instructions for Continued Airworthiness HCC201375000-I/PFA-1</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The use of approved data is subjected to permission of TCCA Services Ltd. Modification Data Summary F111050 Revision N/C.</td>
<td>15/05/2007</td>
<td>Active</td>
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<td>EASA.IM.A.S.02266</td>
<td>HOLLINGSEAD INTERNATIONAL</td>
<td>FAA STC ST1876A</td>
<td>BOEING 747-400 SERIES</td>
<td>Installation of B/E Aerospace Air Chiller PN 470-1</td>
<td>B/E Aerospace Master Data List EC470-9, Revision C, dated November 6, 2006, or later approved revisions</td>
<td>This approval is limited to airplanes on which C&amp;D Zodiac Parity P/N CAB1001-101 has been installed in accordance with STC ST1180A</td>
<td>23/01/2007</td>
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<td>EASA.IM.A.S.02267</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST64040CH-T</td>
<td>GULFSTREAM GV-SP</td>
<td>Completion and installation of the custom interior on Aircraft Serial 5124</td>
<td>In accordance with Gulfstream Aerospace Corporation Master Drawing List GDS2 2048051, Revision B, dated January 30, 2007, or later approved revision FAA approved Flight Manual Supplement, GDS2 2048057, dated February 1, 2007, or later approved revisions.</td>
<td>Aircraft Serial 5124 only. The following STCs are required to be installed as part of this STC: FAA STC ST58389AT-D, EASA Approved EASA.IM.A.S.02147FAA STC ST02796AT-D, EASA Approved FAA STC ST02696AT-D, EASA Approved FAA STC ST02486CH-D, FAA Approved EASA.IM.A.S.01259FAA STC ST04098AT-D</td>
<td>19/02/2007</td>
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<td>EASA.IM.A.S.02275</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SA08315E</td>
<td>CESSNA P172D CESSNA 172SD O CESSNA 172RG CESSNA 172A-S</td>
<td>Installation of vortex generators on the wings and tail surfaces in accordance with the provisions of AML SA08315E.</td>
<td>FAA approved revision Master Drawing List No. MA2139 Rev R5/S3/00 and Installation Instructions MA2139 Rev R5 dated 5/22/00.</td>
<td>Approval of this change in type design applies to the aircraft listed on AML SA08315E only. This approval should not be extended to other aircraft on which other previously approved modification are incorporated unless it is determined that the relationship between this change and any other previous...</td>
<td>13/04/2007</td>
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<td>EASA.IM.A.S.02276</td>
<td>HONEYWELL INTERNATIONAL Inc.</td>
<td>FAA STC SA0686NM</td>
<td>BOEING 787-300</td>
<td>Installation of Honeywell/Racal Aeronautical SATCOM</td>
<td>Master Data List (MDL) ML4078145, Revision AE, dated December 19th, 2006</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>21/12/2006</td>
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<td>EASA.IM.A.S.02277</td>
<td>CENTRILUBE TECHNOLOGIES, LLC</td>
<td>FAA STC SA1435LA</td>
<td></td>
<td>Installation of engines, modified with a Firewire Forward Technologies, LLC modified camshaft, into aircraft listed on the EASA approved ATL</td>
<td>FAA approved Firewire Forward Technologies, LLC Installation Instructions FFT-NP-001-A/FR, revision A, dated 06-Feb-2003, or later approved revision FAA Approved Flight Manual Supplement, P/N 891336, Revision 2, dated 01-Feb-2001, or later...</td>
<td>1. This installation must be done concurrently with EASA STC No. EASA.IM.E.S.01010, which modifies the engine. 2. Regardless of the EASA Approved Type List (ATL), installation is only approved if the Airplane Type / Variant is approved by EASA approved. 3. This STC is approved only for the product...</td>
<td>12/02/2007</td>
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<td>EASA/IM.A.S.02290</td>
<td>ELISEN TECHNOLOGIES</td>
<td>TCAC STC SA07-6</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Main wheel stowage</td>
<td>FAA STC ST00222BO dated 22 November 2005 or later approved revision, Master Data List BAS 63017060 Revision NC dated 31 July 2006 or later approved revision, FAA STC ST0022280 dated 22 November 2005 or later approved revision</td>
<td>FAA STC ST0022280 dated 22 November 2005 or later approved revision, Master Data List BAS 63017060 Revision NC dated 31 July 2006 or later approved revision</td>
<td>31/05/2007</td>
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<td>EASA/IM.A.S.02292</td>
<td>BOMBARDIER INC.</td>
<td>TCAC-LSA07-049D ISSUE 1</td>
<td>BOMBARDIER BD-100-1A10 (CHASE</td>
<td>AIRCRAFT COMPLETION, 10 PASSENGER INTERIOR CONFIGURATION. EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA07-049D Issue 1 dated 19th February 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F017805 Revision</td>
<td>Transport Canada Limited Supplemental Type Certificate C-LSA07-049D Issue 1 dated 19th February 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F017805 Revision</td>
<td>28/02/2007</td>
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<td>EASA/IM.A.S.02293</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA09160AC-D</td>
<td>PIPER PA-32R-300 PIPER PA-32-260 PIPER PA-32-260</td>
<td>Installation of S-TEC System S55/SX Two Axes Automatic Flight Guidance System, Model ST-640</td>
<td>FAA STC ST0022280 dated 22 November 2005 or later approved revision, Master Data List BAS 63017060 Revision NC dated 31 July 2006 or later approved revision, FAA STC ST0022280 dated 22 November 2005 or later approved revision</td>
<td>FAA STC ST0022280 dated 22 November 2005 or later approved revision, Master Data List BAS 63017060 Revision NC dated 31 July 2006 or later approved revision</td>
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<td>EASA/IM.A.S.02294</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC NO. ST3159CH-D</td>
<td>BAE 125 SERIES 800A</td>
<td>Modification of the existing Collins TDR-94D Transponders and Collins RTU-870 Radio Tuning Units to Enhanced Surveillance capable equipment and aircraft interface requirements.</td>
<td>Modification of the existing Collins TDR-94D Transponders and Collins RTU-870 Radio Tuning Units to Enhanced Surveillance capable equipment and aircraft interface requirements.</td>
<td>20/08/2007</td>
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<td>EASA/IM.A.S.02295</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC NO. ST02370CH-D</td>
<td>HAWKER 800XP HAWKER 800 BAE 125 SERIES 800A</td>
<td>Upgrade of the existing dual Universal Avionics Systems Corporation UNS-1D Flight Management System to UNS-1D+ Flight Management System.</td>
<td>Upgrade of the existing dual Universal Avionics Systems Corporation UNS-1D Flight Management System to UNS-1D+ Flight Management System.</td>
<td>31/05/2007</td>
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<td>EASA/IM.A.S.02296</td>
<td>LEARJET Inc.</td>
<td>FAA ST022280</td>
<td>BOMBARDIER CL-660-2B16</td>
<td>Installation of enhanced surveillance modification to Dual Collins TDR-94D transponder system.</td>
<td>Installation of enhanced surveillance modification to Dual Collins TDR-94D transponder system.</td>
<td>31/05/2007</td>
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<td>EASA:IMA.S.02291</td>
<td>EMS TECHNOLOGIES CANADA, Ltd.</td>
<td>TCCA STC-1 SAUS-12 ISSUE 5, DAT</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Installation of EMS HS6-D4I:128I-X/400-400I-HD-710 High Speed Data Transceiver</td>
<td>EMS HD4I-M1298-X/400-400I-HD-710 High Speed Data Transceiver, Engineering Report RTP/4621 by Avionic Services Ltd</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>21/06/2007</td>
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<td>EASA:IMA.S.02299</td>
<td>CASCADE AVIATION SERVICES INC.</td>
<td>STC ST01181SE</td>
<td>DORNIER 328-300 DORNIER 328-100</td>
<td>Installation of increased security cockpit door</td>
<td>Cascade Aviation, Inc. Master Drawing List (MDL) 02CAS096-MDL, Revision D, dated July 14, 2003 or later approved revision</td>
<td>A copy of this Certificate and Cascade Aviation Services Master Drawing List (MDL) 02CAS096-MDL, Revision D, dated July/14, 2003 or later approved revision must be maintained as part of the permanent records for the modified aircraft.</td>
<td>19/01/2007</td>
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<td>EASA:IMA.S.02300</td>
<td>DUNCAN AVIATION, INC.</td>
<td>US STC NO. ST02317CH-D</td>
<td>HAWKER 800 XP HAWKER 800, 800A</td>
<td>Installation of Aircell Iridium ST-3100 single channel SATCOM system.</td>
<td>Duncan Aviation Master Drawing List 041008020, Revision A, dated December 22, 2006 or later FAA approved revision.</td>
<td>Compatibility of this design change with previously approved modifications must be determined by the installer if the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.</td>
<td>31/05/2007</td>
<td>Active</td>
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<td>EASA:IMA.S.02302</td>
<td>DECA AVIATION ENGINEERING LIMITED</td>
<td>TCCA STC-0 LSAS-0-397D ISS 2</td>
<td>A320-214</td>
<td>Installation Of 180 Passenger - All Economy Class Seating Arrangement, in accordance with Transport Canada STC OLSAS-0-397D Issue 2</td>
<td>Master Drawing List, Document Number MS0001 Revision 1, dated 18th July 2002 or later EASA approved revision.Instructions for Continued Airwors, ess, Document Number KCA0071, Revision 1, dated 18th July 2002 or later EASA approved revision.1. As per Transport Canada STC OLSAS-0-397D Issue 2.2. Installation of Iridium ST-3100 SATCOM system in accordance with this STC, or. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft/engine configurations shall be determined by...</td>
<td>10/05/2007</td>
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<td>1011612</td>
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<td>EASA:IMA.S.02306</td>
<td>HONEYWELL INTERNATIONAL Inc.</td>
<td>FAA STC SA473NM</td>
<td>BOEING 747-100-400</td>
<td>Installation of Honeywell/Racal Aeronautical Satellite Communication system</td>
<td>Honeywell Master Data List, Engineering Specification No. ML40751, Revision B, or later approved revisions of this specification.</td>
<td>Approval of this type design applies only to Boeing 747 series aircraft when</td>
<td>10/04/2007</td>
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<td>Model(s)</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>EASA.IMA.S.02309</td>
<td>ENGINE TECHNOLOGIES, Inc.</td>
<td>FAA STD-SA10588SC</td>
<td>CIRRUS SR22</td>
<td>Validation of FAA STC SA10588SC &quot;Installation of modified Teledyne Continental Motors IO-550-A engine equipped with turbosnormalizing system&quot;</td>
<td>Installation in accordance with Engine Technologies Master Drawing List ETH-G2-1, Rev. NC, dated 1 Nov 2006 or later FAA approved Revision</td>
<td>Installation of STC SE10589SC (EASA.IM.E.S.01013) is required for oxygen system meeting requirements of CS23.1441. This STC must be installed in conjunction with this STC. The installer must determine, if this design is compatible with previously approved modifications. If the holder agrees to per...</td>
<td>11/05/2007</td>
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<td>EASA.IMA.S.02312</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC NO. C. - LSA07-154D</td>
<td>BOMBARDIER BD-100-1A10 (CHALLENGER 350)</td>
<td>AIRCRAFT COMPLETION: 10 PASSENGER INTERIOR CONFIGURATION: EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA07-154D Issue 1, dated 15th June 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary F311080 Revision N/C.</td>
<td>1. This STC is limited to aircraft Bombardier BD-100-1A10 Serial Number 5655 or later approved revision.</td>
<td>03/07/2007</td>
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<td>EASA.IMA.S.02314</td>
<td>TDG AEROSPACE INC</td>
<td>REV. 1</td>
<td>FAA STC ST01956LA</td>
<td>BOEING 757-300, 757-200</td>
<td>Universal Fault Interrupter (UFI) Installation</td>
<td>TDG Aerospace MDE 10-3200-0009 (Rev C, 10th January 2007)</td>
<td>The UFI is a Critical Design Control Configuration Limitation (CDCCL) item. Its operation and instructions for maintenance and inspection must be performed in accordance with the TDG Instructions for Continued Airworthiness (document 20-3200-0001, rev. D, 11th January 2007) and must be incorporated into the...</td>
<td>19/02/2007</td>
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<td>1011621</td>
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<td>EASA.IMA.S.02315</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>GULFSTREAM GIV-X</td>
<td>Airplane Completion on Aircraft Serial Number 4066.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GCA1 4914066, Revision D, dated 8 March 2007, or later approved revision.</td>
<td>Aircraft Serial Number 4066 Only. The following STC’s are a required part of this STC: ST29757AT-D, ST29757AT-D, ST29867AT-D, ST29867AT-D, ST29867AT-D, ST29867AT-D and ST30297AT-D. It is the installer's responsibility to ensure these STC’s are all EASA approved.</td>
<td>28/03/2007</td>
<td>Active</td>
<td>1011622</td>
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<td>EASA.IMA.S.02317</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC ST03031CH</td>
<td>BOEING 767-200, 300 SERIES</td>
<td>Installation of Enhanced Mode-S Transponder.</td>
<td>Electronic Cable Specialists Master Data List, Document No. ECS-202463, Revision F, dated June 24, 2006 or later approved revision.</td>
<td>The Airplane Flight Manual Supplement as listed in Appendix A of Electronic Cable Specialists Master Data List, Document No. ECS-202463, Revision F, dated June 27, 2006, or later approved revision is required on board the modified aircraft.</td>
<td>14/06/2007</td>
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<td>EASA STC No.</td>
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<td>FAA STC/STN/STC ref.</td>
<td>Model</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.IM.A.S.02318</td>
<td>VIDEO TECHNOLOGY SERVICES, Inc.</td>
<td>FAA STC ST1956NY</td>
<td>BOEING 767-300</td>
<td>1. Configuration 1: Installation of LCD Monitor System 2. Configuration 2: Installation of LCD Monitor System</td>
<td>1. Certification Basis 1: in accordance with Video Technology Services, Inc. Technical Drawing List Document No. 205-D-334, Revision B, dated August 15, 2004. 2. Configuration 2: in accordance with Video Technology Services, Inc. Technical Drawing List Document No. 205-D-368, Revision C, dated June 2, 2006.</td>
<td>1. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give notice to the other person written evidence of that permission. 2. The aircraft must be maintained in accordance with Video Technology Services, Inc. Instructions for Continued Airworthiness...</td>
<td></td>
<td>03/06/2007</td>
<td>Active</td>
<td>101/1623</td>
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<tr>
<td>EASA.IM.A.S.02321</td>
<td>HAWKER BEECHCRAFT CORPORATION (HBC)</td>
<td>FAA STC SA1417WH</td>
<td>RAYTHEON A36</td>
<td>Replace pilot's altimeter with IS&amp;S encoding altimeter.</td>
<td>1. RAS Master Data List Replace RAS23060940, Revision 6, dated December 7, 2006, or later approved revisions. 2. Airplane/Flight Manual Supplement RAS23060958, dated December 12, 2006, or later approved revisions.</td>
<td>Compliance with the following requirements, in addition to the requirements shown on the Type Certificate Data Sheet 3A20, has been shown: 23.303, 23.305, 23.307, 23.321, 23.341, 23.365, 23.601, 23.609, 23.627 effective February 1, 1996, 2011 as amended through Amendment 23-723-607 as amended through...</td>
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<td>26/02/2007</td>
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<td>101/1627</td>
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<td>EASA.IM.A.S.02322</td>
<td>MCCAULEY PROPELLER SYSTEMS</td>
<td>FAA STC SA66CH</td>
<td>BEECH 23</td>
<td>Installation of McCauley Propeller Model B3D36C429/82NPR-6</td>
<td>In accordance with McCauley Technical Report No. 844, no revision, dated January 15, 2004 and McCauley Drawing No. D-69195, No revision, dated April 1, 1998, or later approved revision FAA approved Flight Manual Supplement, Technical Report No. 845, dated April 3, 1998, or later approved revision...</td>
<td>1. Aircraft Specifications are the same as listed on Type Certification Data Sheet AIC, except as follows: Propeller and Propeller Limits Propeller: McCauley Constant Speed B3D36C429/82NPR-6 Number of blades: 3Blade Angles: At 3° station: Low 10.5 degrees, High 33.8 degrees Diameter: 76...</td>
<td></td>
<td>22/05/2007</td>
<td>Active</td>
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<td>EASA.IM.A.S.02324</td>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD.</td>
<td>TCCA STC Q-LSA06-018D</td>
<td>BOMBERDIER BD-700-1A10</td>
<td>Re-locate Electronic Flight Bag and Miscellaneous Cabin Modifications in Accordance of SN 9455/Validation of TCCA Limited STC Q-LSA06-018D to SN 9456/MDL 26115002 Rev A, Pilot's FAA STC Q-LSA06-018/D.</td>
<td>IFC-3A, TL26115-001 Rev 5, or later TCCA approved revisions</td>
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<td>22/03/2007</td>
<td>Active</td>
<td>101/1630</td>
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<td>EASA.IM.A.S.02325</td>
<td>RAISBECK ENGINEERING Inc.</td>
<td>STC ST1724SE DATED JANUARY ST</td>
<td>LEARJET 31</td>
<td>Installation of a prop reduction system (contoured flaps).</td>
<td>AFM Supplement 08L-5786, revision IR (or later approved revisions).</td>
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<td>22/11/2007</td>
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<td>EASA.IM.A.S.02326</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA SA85-67</td>
<td>DHC-8-300 DHC-8-100, DHC-8-200</td>
<td>Installation of Aero Consulting Services Ltd. &quot;Improved Exhaust Outlet Shroud Assembly&quot; according to Transport Canada STC SA 95-87</td>
<td>The manufacture and installation of the Improved Exhaust Outlet Shroud Assembly is to be completed in accordance with Aero Consulting Services Ltd. Modification Data Summary Number A23820, Revision D, dated March 21, 2001 and later DOT approved revision. Required equipment: Aero consulting Services Lt...</td>
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<td>01/03/2007</td>
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<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>FAA STC ST97669C-D L-3 COMMUNICATIONS INTEGRATED SYSTEMS TEM LPT</td>
<td>787</td>
<td>BOEING 787</td>
<td>Installation of a First-Class Interior/The project installs all first-class interior in a Swiss registered Boeing 787 aircraft. This interior consists of installing 54 seats, a bar and dining area, and upgrades to entertainment system electronics.</td>
<td>FAA STC ST01732SEL - L-3 Communications Integrated Systems L.P. Drawing List Section 2.1 of Report 33263-0004, Revision A, dated May 1, 2007.</td>
<td>1. Instruction for Continued Amorteness (ICA) dated May 1, 2007.2. This approval is limited to only the installation made in the Boeing Model 787-308, Serial Number 30393 aircraft.</td>
<td>15/02/2007</td>
<td>Active</td>
<td>10/16132</td>
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<td>FAA STC ST101275BD-7675-0001-00, Rev C, Drawing List (MDL) No. AFMS CEB1 349M000</td>
<td>BOMBERDIER COMPLETION CENTRE INC</td>
<td>BD-700-1A10</td>
<td>Customized Interior Completion - SN 9239 installation of TCJA STC C-LSA17-0370 Issue 1</td>
<td>AFM Supplement G- FM25000155,TCCA SoC</td>
<td>Refer to the Limitations sections of the required FM Supplement,WLAN system is compatible to IEEE 802.11 (b). The use of WLAN is subject to operational rules.Prior to installation ofthis modification the installer must determine that the internrelationship between this modification and any other prev...</td>
<td>09/07/2007</td>
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<td>FAA STC NO. 767-200, 767-300, 767-400ER</td>
<td>AIRCRAFT SYSTEMS AND MANUFACTURING G. INC</td>
<td>ST019065C</td>
<td>Installation of Mode S Upgrade, Elementary and Enhanced Surveillance Parameters</td>
<td>ASI Top Drawing List No. 5675-0001-00, Rev C, 06/12/2006</td>
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<td>FAA STC ST02873AT-D REV. 1 GULFSTREAM AEROSPACE CORPORATION</td>
<td>GULFSTREAM 200</td>
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<td>Installation of a Runway Awareness and Advisory System i.e. Index List CEB1 3498000 Note: This Revision corrects a typo in the Original Product Type Certificate Number.</td>
<td>AFMS CEB1 3498000, AFMS CEB1 3498000 Rev.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previos installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by th...</td>
<td>04/03/2009</td>
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<td>10/16138</td>
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<td>FAA STC ST01732SE LITEAR AVIATION PRODUCTS Inc.</td>
<td>BOMBERDIER CL-600-2B19</td>
<td>Installation of light weight window plug</td>
<td>FAA STC ST01732SE Rev. II</td>
<td>N/A</td>
<td>N/A</td>
<td>15/03/2007</td>
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<td>FAA STC ST02883AT-D</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>GULFSTREAM 200</td>
<td>Installation of Transponder Flight ID i.e. Index List CEB1 1268000, Rev. NC or later FAA approved version,</td>
<td>AFMS CEB1 1268000 for -501 and -503 configuration; AFMS CEB1 1268000 for -505 configuration;</td>
<td>20/08/2007</td>
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<td>Limitations/Conditions</td>
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<td>EASA.IM.A.S.02338</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST038AT-D</td>
<td>GULFSTREAM 200</td>
<td>Installation of Universal UNS 1C+ Dual FMS Software upgrade a.e. Index list CEB1 118800 Rev. 8 or later approved version.</td>
<td>AFMS CEB1 118M003</td>
<td></td>
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<td>Compatibility of this design change with previously approved modifications must be determined by the installer. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.</td>
<td>20/08/2007</td>
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<td>EASA.IM.A.S.02339</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST087AT-D</td>
<td>GULFSTREAM IV</td>
<td>Installation of Gabrius Flight Identification. In accordance with Gulfstream Aerospace Corporation Index List CEB1 118M000, Revision NC, dated June 11, 2004, or later FAA approved revision is required for the -501 config.</td>
<td>AFMS CEB1 118M003</td>
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<td>19/10/2007</td>
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<td>EASA.IM.A.S.02345</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0228CH-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of the Heads-Up Technologies CMS-400 Checklist Management System. Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514258000, Revision C dated November 28, 2007, or later FAA Approved revision. Operation to be in accordance with Gulfstream Flight Manual Supplement GC414225M000, Revision B, dated November 28, 2007, or later approved revisions.</td>
<td>AFMS CEB1 118M003</td>
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<td>EASA.IM.A.S.02346</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST01911A-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Aircraft Serial No. 5139 completion and cabin interior installation.</td>
<td>AFMS CEB1 118M003</td>
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<td>14/06/2007</td>
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<td>EASA.IM.A.S.02348</td>
<td>AVIATION TECHNICAL SERVICES, Inc.</td>
<td>FAA ST003NM-D</td>
<td>737-300</td>
<td>Installation of Allied Signal p/n 967-0102-001 Digital Flight Data Acquisition Unit (DFDAU) and necessary sensors and wiring to record Twenty Two (22) Parameters of data per Tramco Incorporated Master Drawing List 95-180 Sheet 1 Revision A dated 12 January, 1996 or later FAA/DAS approved revision. - Tramco Incorporated Master Drawing List 95-180 Sheet 1 Revision A, dated January 12, 1996 or later EASA approved revision.</td>
<td>AFMS CEB1 118M003</td>
<td></td>
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<td>1. Approval of this change in Type Design applies to the above model aircraft only. This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated unless it is determined that the relationship between this change and any oth...</td>
<td>27/02/2007</td>
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<td>EASA STC no.</td>
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<td>FAA STC No.</td>
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<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.02340</td>
<td>LEARJET Inc.</td>
<td>FAA STC No.</td>
<td>BOMBARDIER CL-600-2B16 (CL-604</td>
<td>Replacement of a Collins TCAS II bottom OMNI-Directional Antenna (Sensor Systems)</td>
<td>FAA STC No.</td>
<td>09/08/2007</td>
<td>Active</td>
<td>10016140</td>
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<td>EASA.IM.A.S.02351</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC NO.</td>
<td>HAWKER 800 XP</td>
<td>Replacement of Existing 3, 4 and 5 auxiliary batteries with Securaplane XL245 lead acid batteries</td>
<td>Duncan Aviation Master Drawing List 00412006, Revision C, dated December 20, 2006 or later approved revision</td>
<td>1. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.</td>
<td>30/04/2007</td>
<td>Active</td>
<td>10016146</td>
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<td>EASA.IM.A.S.02352</td>
<td>ASSOCIATED AIR CENTER, L.P.</td>
<td>FAA ST10536SC</td>
<td>A319-115</td>
<td>Installation of an Executive Style Interior in accordance with Master Document List, Document Number 245900000, Revision R, dated 25 October 2007, or later EASA approved revision</td>
<td>AFM Supplement, Document Number 245902100, Revision R, dated 25 October 2007, or later EASA approved revision.</td>
<td>The aircraft is limited to Private use only (not to be operated for hire, reward or common carriage). This STC is approved only for the product configuration as defined in the above Master Document List. Limitations and Conditions specified in FAA STC ST10536SC D remain applicable.</td>
<td>30/10/2007</td>
<td>Active</td>
<td>10016150</td>
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<tr>
<td>EASA.IM.A.S.02353</td>
<td>RAYTHEON AIRCRAFT SERVICES, INC.</td>
<td>FAA STC</td>
<td>400, 400A, 400T</td>
<td>Replacement of a Standby Altimeter installation in the Raytheon (Beech) models 400, 400A, 400T aircraft inaccordance with Master Data List (MDL), RA51106DL004, Revision 1, dated, January 4, 2007, or later FAA Approved Revisions.</td>
<td>see above</td>
<td>This approval should not be extended to other aircraft of this model on which other previously approved modifications are incorporated unless it is determined by the installer that the interrelationship between this change and any of those other previously approved modifications will produce no adverse...</td>
<td>25/04/2007</td>
<td>Active</td>
<td>10016151</td>
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<td>EASA.IM.A.S.02354</td>
<td>AEROCET, INC.</td>
<td>FAA STC</td>
<td>180A, 180H, 18U, 180K</td>
<td>Installation of Aerocet Model 3500 or 3500L seaplane floatson Cessna 180 Aircraft.</td>
<td>Installation must be in accordance with Master Document List 35-10006 Rev. B dated 21 June 1994 or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement No. A-10018 (180A-180F), A-10019 (180G-180M) and A-10020 (180J, 180K dated 21 October 1994, or later approved revision...</td>
<td>Those Cessna Model 180G, 180H, 18U and 180K airplanes equipped with the structural modifications defined in Kamero AirHarbour (STC) No. SA649NW, Amendment dated 14 July 1984, may be operated on Aerocet Model 3500 or 3500L Boats at gross weight up to 3190lbs, subject to the limitation specified in A...</td>
<td>09/08/2007</td>
<td>Active</td>
<td>10016152</td>
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<tr>
<td>EASA.IM.A.S.02355</td>
<td>AEROCET, INC.</td>
<td>FAA STC</td>
<td>CESSNA A18SF CESSNA A185E CESSNA A185E</td>
<td>Installation of Aerocet Model 3500 or 3500L seaplane floatson Cessna 185 Aircraft.</td>
<td>Installation must be in accordance with Master Document List 35-10006 Rev. B dated 21 June 1994 or later Approved Revisions. Operation must be in accordance with Flight Manual Supplement No. A-10018 (180A-180F), A-10019 (180G-180M) and A-10020 (180J, 180K dated 21 October 1994, or later approved revision...</td>
<td>The airplane must incorporate the Cessna approved seaplane kit referred to Section 7 of the Aerocet AFM. The airplane must be equipped with the Cessna rudder centering device/non-linear rudder centering spring system defined on Aerocet Drawing No. 35-10000, Sheet 5. The installation of this modifi...</td>
<td>09/08/2007</td>
<td>Active</td>
<td>10016153</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<th>EASA STC No.</th>
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<th>FAA STC</th>
<th>Description</th>
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<tr>
<td>EASA.IM.A.S.02359</td>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA SA19446A</td>
<td>BEECH F90 BEECH F60 BEECH F50 BEECH B200T BEECH B200C BEECH B200 BEECH A200 BEECH A200C BEECH A200 BEECH 200T BEECH 200CT BEECH 200 BEECH 200</td>
<td>Installation of Ultra Electronics Engine Instruments Revision 1. Installation of powerplant indication system. This project is to amend EASA of EASA STC No. EASA.IM.A.S.02359. - Installation of in accordance with Master Drawing List No. 90-000, Rev. C, dated 24 July 2008 or later approved revision. Operation must be in accordance with Aircraft Flight Manual Supplement No. AFMS 55023-11 Rev B, dated 30 Sept 2008 or later approved revision. Maintenance must be in accordance with Aircraft Flight Manual Supplement No. AT 10011, Revision 1, dated 26-Jan-2006, or later approved revision.</td>
<td>- Installation of in accordance with Master Drawing List No. 90-000, Rev. C, dated 24 July 2008 or later approved revision. Operation must be in accordance with Aircraft Flight Manual Supplement No. AFMS 55023-11 Rev B, dated 30 Sept 2008 or later approved revision. Maintenance must be in accordance with Aircraft Flight Manual Supplement No. AT 10011, Revision 1, dated 26-Jan-2006, or later approved revision.</td>
<td>27/05/2009</td>
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<tr>
<td>EASA.IM.A.S.02377</td>
<td>SIMMONDS PRECISION PRODUCTS, Inc</td>
<td>FAA STC ST0021660</td>
<td>BOEING 737-400</td>
<td>Installation of the Goodrich Corporation Fire Detection (Class &quot;E&quot;)</td>
<td>Goodrich Corporation Master Drawing List T3078- MDL-2001, Revision H, dated May 30, 2006 or later approved revision.</td>
<td>07/05/2007</td>
<td>Active</td>
<td>10016156</td>
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<tr>
<td>EASA.IM.A.S.02381</td>
<td>AIRCRAFT TECHNICAL SERVICE, Inc.</td>
<td>FAA STC SA02123LA</td>
<td>200</td>
<td>Installation of a nose mounting truss, bulkhead, camera, electronic backpack and missile seeker head.</td>
<td>Installation in accordance with Aircraft Technical Service Aircraft Data List 3500, Revision H, dated March 12, 2008, or later approved revisions. Aircraft to be operated in accordance with FAA Approved Aircraft Flight Manual Supplement ref Document No ATS Report 2734, RevisionNew, dated March 27...</td>
<td>28/04/2008</td>
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<td>EASA STC no.</td>
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<td>Certification basis</td>
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<td>Limitations/Conditions</td>
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<td>EASA.IM.A.S.02382</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01082WI-D</td>
<td>LEARJET 60</td>
<td>Installation of 20&quot;, 312&quot; or 376&quot; thick one piece windshields (two piece windshield to one piece windshield conversion).</td>
<td>FAA STC ST01082WI-D</td>
<td>Design Compliance Report 60-D1810 ’MDL M10003/80/2/C’ or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>03/08/2007</td>
<td>Active</td>
<td>10016160</td>
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<tr>
<td>EASA.IM.A.S.02385</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST02349CH-D</td>
<td>GULFSTREAM IV</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 1331.</td>
<td>Gulfstream Aerospace Report CE428043002, Revision NC. And Master Drawing List CE428048000, Revision B, dated May 1, 2007 or later approved revisions. Operate in accordance with Gulfstream Flight Manual Supplement CE42 804M000, Revision -, dated May 3, 2007 or later approved revisions.</td>
<td>31/05/2007</td>
<td>Active</td>
<td>10016162</td>
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<td>EASA.IM.A.S.02390</td>
<td>LEARJET Inc.</td>
<td>FAA STC SAA100NM-D</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Installation of an Axle Jack Storage Box Installation</td>
<td>DAS-12-M1 Certification Plan/CA-3074-TUC Master Drawing List/PO0-C-07 Compliance Checklist / MOCSTR-M101 Structures Report</td>
<td>Aircraft S/N 1331 Only.</td>
<td>31/05/2007</td>
<td>Active</td>
<td>10016163</td>
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<tr>
<td>EASA.IM.A.S.02391</td>
<td>LEARJET Inc.</td>
<td>FAA STC SAA097NM-D</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Installation of an Engine Cover Storage - All Equipment</td>
<td>DAS-12-M1 Certification Plan/CA-3074-TUC Master Drawing List/PO0-C-07 Compliance Checklist / MOCSTR-M111 Structures Report</td>
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<td>31/05/2007</td>
<td>Active</td>
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<tr>
<td>EASA.IM.A.S.02392</td>
<td>LEARJET Inc.</td>
<td>FAA STC SAA003NM-D</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Interior Installation/Modification in accordance with DAS approved Learjet Master document list No. CA-15007/TUC, Rev. 5’dated January 12, 1996 or later approved revision</td>
<td>Learjet Airplane Manual Supplement C1002, original release, dated March 5, 1993; C1003 original release, dated Fohr 23, 1993; C1004, original release, dated Fohr 23, 1993; C1005, original release, dated March 10, 1993; C1011, original release, dated Oct 12, 1993 or later approved revision</td>
<td>The proper interrelationship between this installation and any previously approved configuration has to be assured.</td>
<td>03/08/2007</td>
<td>Active</td>
<td>10016165</td>
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<td>EASA STC no.</td>
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<td>Associated Technical Documentation</td>
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<td>EASA/IMA.S.02393</td>
<td>RAYTHEON-aircraft services, inc.</td>
<td>FAA STD- SA01366W</td>
<td>BEECH 300</td>
<td>Modification to install a Collins IP5500, a file server unit, cursor control panel, XM weather receiver and antenna</td>
<td>RAS Master Data List RAS050500169, Rev 2.0, dated 20 June 2006 or later FAA approved Revision FAA approved RAS AFM 5 Doc. No. RAS050501083, dated 7 July 2006, or later FAA approved Revision</td>
<td>see above</td>
<td>03/07/2007</td>
<td>Active</td>
<td>10116167</td>
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<tr>
<td>EASA/IMA.S.02394</td>
<td>RAYTHEON-aircraft services, inc.</td>
<td>FAA STC- SA01426W</td>
<td>C90A, C90GT</td>
<td>Installation of replacement static inverters. Data required: 1. RAS Master Data List RAS230600970, Revision 1.0, dated January 30, 2007, or later approved revision.</td>
<td>ICA-29R2-706 revision NC or later approved revision Instruction for continued airworthiness ICA-29R2-706 revision NC or later approved revision.</td>
<td>03/05/2007</td>
<td>Active</td>
<td>10116167</td>
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<td>EASA/IMA.S.02395</td>
<td>ELISEN-technologies</td>
<td>TCCA STC- SA07-20</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Miscellaneous Storage Box, Project No. 29R2-707</td>
<td>MDL-29R2-707 rev NC or later approved revision Instruction for continued airworthiness ICA-29R2-707 revision NC or later approved revision.</td>
<td>03/07/2007</td>
<td>Active</td>
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<td>EASA/IMA.S.02396</td>
<td>ELISEN-technologies</td>
<td>TCCA STC- SA07-21</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Towbar Stowage, Project No. 29R2-708.</td>
<td>MDL-29R2-708 revision NC or later approved revision. Instructions for continued airworthiness ICA-29R2-708 revision NC or later approved revision.</td>
<td>05/07/2007</td>
<td>Active</td>
<td>10116169</td>
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<td>EASA/IMA.S.02397</td>
<td>ELISEN-technologies</td>
<td>TCCA STC- SA07-22</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Equipment Bay Forward Catwalk, Project No. 29R2-709.</td>
<td>MDL-29R2-709 revision NC or later approved revision. Instructions for continued airworthiness ICA-29R2-709 revision NC or later approved revision.</td>
<td>03/07/2007</td>
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<td>10116170</td>
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<td>EASA/IMA.S.02398</td>
<td>ELISEN-technologies</td>
<td>TCCA STC- SA07-23</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Equipment Bay Stowage Box, Project No. 29R2-710.</td>
<td>MDL-29R2-710 revision NC or later approved revision. Instructions for continued airworthiness ICA-29R2-710 revision NC or later approved revision.</td>
<td>04/07/2007</td>
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<td>10116171</td>
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<td>EASA/IMA.S.02399</td>
<td>BOMBARDIER-aircraft services, inc.</td>
<td>TCCA C- LSA07-204D</td>
<td>BD-100-1A10 (CHALLENGER 300)</td>
<td>AIRCRAFT COMPLETION, 9 PASSENGERS INTERIOR CONFIGURATION ICA Certification of Transport Canada Limited Supplemental Type Certificate C-LSA07-204D Issue 1, dated 7th September 2007, in accordance with Aero ConsultingServices Ltd. Modification Data Summary F384000 Revision N/C</td>
<td>Transport Canada Limited Supplemental Type Certificate C-LSA07-204D Issue 1, dated 7th September 2007. • Transport Canada Approved Aero Consulting Services Ltd. Aircraft Flight Manual Supplement F384000 Revision N/C or later Transport Canada approved revision. Instructions for Continued...</td>
<td>24/09/2007</td>
<td>Active</td>
<td>10116172</td>
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<tr>
<td>EASA/IMA.S.02400</td>
<td>HONEYWELL-international inc.</td>
<td>FAA STC- ST00717W</td>
<td>BOEING 737-400 BOEING 737-300 BOEING 737-500 BOEING 737-600 SERIES</td>
<td>Installation of the Honeywell Mark II ACARS Master Drawing List 159-03189-0001, Revision J, dated July 25, 2004, or later approved revision.</td>
<td>Master Drawing List 159-03189-0001, Revision J, dated July 25, 2004, or later approved revision. Airplane Flight Manual Supplement 796-0182-800 revision &quot;Original&quot;, dated September 16, 2004, or later approved revision. Compatibility of this design change with previously approved modifications must be determined by the installer. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.</td>
<td>21/05/2007</td>
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<td>10116173</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>STC Holder: GULFSTREAM 10016179, 10016176, 10016177, 10016182, 10016181</td>
<td><strong>Installation of Traffic Alert and Collision Avoidance System (TCAS II) and GARMIN Transponder</strong>&lt;br&gt;Installation of a Collins Traffic Alert and Collision Avoidance System (TCAS II) Transmitter/Receiver with Honeywell TA/RA/I/CS II.</td>
<td>FAA STC 91-10947</td>
<td>PIPER PA-44-150</td>
<td>Bulkin No. 719 Revision IR, dated 05-Mar-1998, or later approved revision; for field installations: Master Drawing No. 92988; Initial Revision, dated 05-Mar-1998, or later approved revision; for Piper Factory Installations: Master Drawing List No. 92988-1; Initial Revision, dated 30-Mar-1998; or later approved revision(s).</td>
<td>Prior to installation of this modification the installer must ensure that the actual TCAS installation or any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2007</td>
<td>Active</td>
<td>1011872</td>
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<td><strong>Installation of Lower Deck Mobile Rest Area</strong>&lt;br&gt;Installation of a Divan Cargo Net Assembly.</td>
<td>FAA STC 19707</td>
<td>REV. 2</td>
<td>FAA ST09555C-D</td>
<td>HAWKER 900 XP HAWKER 800XP, HAWKER 750, HAWKER 800, SAE.125 SERIES 800A</td>
<td>FAA STC ST0826LA: Max. Flap Extension Speed VFE: 10° Flap 195 mph (CAS) 30° Flap 145 mph (CAS) Min. Single Engine Control Speed VMC: 10° Flap 83 mph (CAS) Flap up (...</td>
<td>20/05/2008</td>
<td>Active</td>
<td>1011879</td>
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<td><strong>Installation of Traffic Alert and Collision Avoidance System (TCAS II) and GARMIN Transponder</strong>&lt;br&gt;Installation of a Collins Traffic Alert and Collision Avoidance System (TCAS II) Transmitter/Receiver with Honeywell TA/RA/I/CS II.</td>
<td>FAA STC 91-10947</td>
<td>PIPER PA-44-150</td>
<td>PIPER PA-44-150</td>
<td>Bulkin No. 719 Revision IR, dated 05-Mar-1998, or later approved revision; for field installations: Master Drawing No. 92988; Initial Revision, dated 05-Mar-1998, or later approved revision; for Piper Factory Installations: Master Drawing List No. 92988-1; Initial Revision, dated 30-Mar-1998; or later approved revision(s).</td>
<td>Prior to installation of this modification the installer must ensure that the actual TCAS installation or any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2007</td>
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<td>1011872</td>
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<td>EASA STC No.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.I.M.A.S.02412</td>
<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>FAA STC ST 0380 AT</td>
<td>SD-60 SHORTS</td>
<td>Installation of a Honeywell A44 V EGPWS.</td>
<td>• FAA STC ST0380 AT-D: MDL SD-60-3443-0001 Rev 1 dated 03/06/05 - AFMS SD-60-3443-1012.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>03/07/2007</td>
<td>Active</td>
<td>10110161</td>
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<td>EASA.I.M.A.S.02415</td>
<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>FAA STC ST 03099 AT</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Installation of a replacement landing gear wiring harness.</td>
<td>FAA STC ST03099AT, N/A</td>
<td>29/05/2007</td>
<td>Active</td>
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<td>EASA.I.M.A.S.02416</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>US FAA STC NO ST01775LA-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of an Avionica Inc. Miniature Quick Access Data Recorder (Mini QAR).</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC51 4208000, Revision C, dated 27 September 2006, or later approved revisions.</td>
<td>21/06/2007</td>
<td>Active</td>
<td>10110167</td>
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<td>EASA.I.M.A.S.02417</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03409AT-D</td>
<td>GULFSTREAM GV-X</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4004.</td>
<td>Aircraft Serial Number 4004 only. The following STCs are a required part of this STC: ST02948AT-DST02979AT-DST02987AT-DST02994AT-D. It is the installer's responsibility to ensure these STCs are all EASA approved.</td>
<td>09/10/2007</td>
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<td>EASA.I.M.A.S.02429</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA SA01-65, ISSUE 4</td>
<td>DHC-8-300 DHC-8-100, DHC-8-200</td>
<td>Additon of filtration to Cabin Recirculation Air. Addition of recirculation air filter reduces potential blockage of the condenser mixer and improves peak performance of the ECS system. TCCA STC SA01-65, Issue 4, dated 24 June 2005.</td>
<td>Aero Consulting Services Modification Data Summary No. C46000 Revision D or later approved revisions for DHC-8-100 and DHC-8-200 series and S46030 Revision C or later TCCA approved revisions for DHC-8-300 series Maintenance Manual Supplement (MMS) C46071 and Parts Manual Supplement (PMS) C46073 for...</td>
<td>06/07/2007</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC No.</td>
<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>EASA/IMA.S.02430</td>
<td>CANARD AEROSPACE CORPORATION</td>
<td></td>
<td>FAA STC ST01106CH</td>
<td>BOEING 747-5S</td>
<td>Installation of Digital Flight Data Recorder</td>
<td>Data package containing the following documents, and additional supporting data 1. Certification Plan 31-3314-01 Rev. B2. Top level Document Control Drawing No. 30010, Rev. G, dated 6 October 2003, or later FAA approved revisions, FAA STC number ST01106CH.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft configurations shall be determined by the installer.</td>
<td>20/10/2007</td>
<td>Active</td>
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<td>EASA/IMA.S.02433</td>
<td>HONEYWELL INTERNATIONAL Inc.</td>
<td></td>
<td>FAA STC ST00361A</td>
<td>GULFSTREAM IV</td>
<td>Installation of the Honeywell HUD-2020 Head up Display System on Gulfstream G-IV aircraft for Category II approach criteria.</td>
<td>In accordance with Master Data List Spec No 5228-0867, Revision G dated July 26, 1997 or later approved revisions.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/02/2009</td>
<td>Active</td>
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<tr>
<td>EASA/IMA.S.02434</td>
<td>S-TEC CORPORATION</td>
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<td>FAA STC SAB9992AC-D</td>
<td>MX-7.1-180, MX-7.1-180A, MT-7-235, MT-7-235A, MX-7.180A, MX-7.180B, MX-7.180C, M-7-260, M-7-260C, M-7-235B, M-7-235C, MX-7-235, M-8-210, M-8-235C, M-8-235</td>
<td>Installation of S-TEC System, Model 25030 Single and Two Axis Automatic Flight Guidance Systems, Model ST-810</td>
<td>FAA approved S-TEC Bulletin No.910, Revision 4, dated 28-Dec-2000, or later approved revision/FAA approved S-TEC Master Drawing List No. S20185, Revision D, dated 28-Dec-2000, or later approved revision/FAA/DAS Approved Pilot’s Operating Handbook and/or Airplane Flight Manual Supplement, PN 891691S.</td>
<td>This STC is eligible on variants M-5-210C, M-6-235, M-6-235C, M-7-180, M-7-235, M-7-235A, M-7-235B, M-7-235C and MX-7-235 in landplane, floatplane, or amphibian configuration This STC is eligible on variants MX-7-180A, MX-7-180B, MX-7-180C, MT-7-180A, MT-7-235A, MT-7-235B, MT-7-235C and MX-7-235 in landplane, floatplane, or amphibian configuration This STC is eligible on variants MX-7-180A, MX-7-180B, MX-7-180C, MT-7-180A, MT-7-235A, MT-7-235B, MT-7-235C and MX-7-235 in landplane configuration...</td>
<td>03/07/2007</td>
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<td>EASA/IMA.S.02437</td>
<td>BOMBARDIER INC.</td>
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<td>TCIA STC C-LSA07-2SBD</td>
<td>BOMBARDIER BD-100-1A10 (CHALIE)</td>
<td>Aircraft Completion, 10 passengers interior configuration EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA07-2SBD, Issue 1, dated 22nd October 2007, in accordance with Aerocan Consulting Services Ltd. Modification Data Summary F494000 Revision A dated 25.10.2007.</td>
<td>Transport Canada Limited Supplemental Type Certificate C-LSA07-2SBD, Issue 1, dated 22nd October 2007, Transport Canada Approved Aero Consulting Services Ltd. Airplane Flight Manual Supplement F494090 Revision N/C or later Transport Canada approved revision/Instruction for continued Airworthiness ref...</td>
<td>This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20155.</td>
<td>31/10/2007</td>
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<td>EASA.IM.A.S.02438</td>
<td>PEMCO WORLD AIR SERVICES</td>
<td>REV. 2</td>
<td>FAA STC SA26685O</td>
<td>Boeing 737-400</td>
<td>Modification of floor in accordance with Pemco World Air Services, Inc. Master Data list, Report No. 2316, revision &quot;K&quot;, dated June 15, 2006, or later EASA approved revision.</td>
<td>- Master Data list, Report No. 2316, revision &quot;K&quot;, dated June 15, 2006, or later EASA approved revision. - Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2008, or later EASA approved revision.</td>
<td>Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2008, or later EASA approved revision, is a required part of this STC. The operator must be incorporated into the operator's maintenance program.</td>
<td>24/06/2008</td>
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<td>EASA.IM.A.S.02439</td>
<td>PEMCO WORLD AIR SERVICES</td>
<td>REV. 2</td>
<td>FAA STC SA26685O</td>
<td>Boeing 737-400</td>
<td>Installation of a cargo door in accordance with Pemco World Air Services, Inc. Master Data list, Report No. 2373, revision &quot;T&quot;, dated June 15, 2006, or later EASA approved revision.</td>
<td>- Master Data list, Report No. 2373, revision &quot;T&quot;, dated June 15, 2006, or later EASA approved revision. - Airplane Flight manual Supplement (AFMS), Report P03-003FM Rev 7, or later EASA approved revision. - Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2008, or later EASA approved revision, is also a required part of this STC.</td>
<td>Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2008, or later EASA approved revision, is a required part of this STC, and should be incorporated into the operator's maintenance program.</td>
<td>24/06/2008</td>
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<td>EASA.IM.A.S.02440</td>
<td>PEMCO WORLD AIR SERVICES</td>
<td>REV. 2</td>
<td>FAA STC SA26685O</td>
<td>Boeing 737-400</td>
<td>Installation of a Class &quot;E&quot; Cargo Compartment and passenger interior configuration in accordance with Pemco World Air Services, Inc. Master Data list, Report No. 2374, revision &quot;S&quot;, dated June 15, 2006, or later EASA approved revision.</td>
<td>- Master Data list, Report No. 2374, revision &quot;S&quot;, dated June 15, 2006, or later EASA approved revision. - Airplane Flight manual Supplement (AFMS), Report P03-003FM Rev 7, or later EASA approved revision. - Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2008, or later EASA approved revision, is also a required part of this STC.</td>
<td>Instructions for Continued Airworthiness (ICA), Pemco Report No 2917, Revision C, dated January 18, 2008, or later EASA approved revision, is also a required part of this STC, and should be incorporated into the operator's maintenance program.</td>
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<td>EASA.IM.A.S.02443</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC SA26685O</td>
<td>BOEING 737-400</td>
<td>Gulfstream GV-SP, S/N 5142 completion and cabin interior installation.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC52 2046058, Rev. A, dated 9 July 2007, or later approved revisions. Operations to be carried out in accordance with Gulfstream Flight Manual Supplement GC52 204M019, original release, dated 17 July 2007, or later approved revisions.</td>
<td>GV-SP S/N 5142 only.</td>
<td>26/07/2007</td>
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<td>EASA.IM.A.S.02444</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC SA1267WI-D</td>
<td>BEECH B300C BEECH B300 BEECH 300</td>
<td>Installation of Collins Pro Line 21 Avionics System.</td>
<td>Installation in accordance with Collins Master Drawing List ref RCO-020-001, Revision C, dated 22 July 2008, or later approved revisions. Operations to be carried out in accordance with Collins Flight Manual Supplement ref RCO-030, Revision C, dated 21 July 2008, or later approved revisions.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect on the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>S-TEC CORPORATION</td>
<td>FAA STC SA0287AC-D</td>
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<td>PIPER PA-28R-201 T</td>
<td>Installation of S-TEC System, Model 2030 Single and Two Axis Automatic Flight Guidance Systems, Model ST-729</td>
<td>• S-TEC Bulletin No.953, dated 08-Oct-1997, Initial Issue or later approved revision • S-TEC Master Drawing List No. 901051, dated 08-Oct-1997, Initial Issue or later approved revision • FAA/DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supplement, P/N BN5141, dated 1...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>26/07/2007</td>
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<td>EASA.IM.A.S.02444</td>
<td>COMMUTER AIR TECHNOLOGY Inc.</td>
<td>FAA STC SA10370SC</td>
<td></td>
<td>E90, H90</td>
<td>Installation of exhaust gas extractors on Pratt &amp; Whitney Canada PT6A-27, -28, -29, -34, -34B, -135, or 135A turboprop engines.</td>
<td>Installation in accordance with Commuter Air Technology Drawing List No. DL00201, Rev. 8, dated October 8, 2004, or later approved revision.</td>
<td>Note that the airspeed and shaft horsepower limits listed in the original Airplane Flight Manuals are still applicable (see TC Datasheet No. 3420). Compatibility of this design change with previously approved modifications must be determined by the installer. If the holder agrees to permit another person to use this certificate to alter the product, the holder agrees to permit another person to use this certificate to alter the product.</td>
<td>21/05/2007</td>
<td>Active</td>
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<td>EASA.IM.A.S.02447</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST01508LA-D</td>
<td>GULFSTREAM GV-5P</td>
<td>Aircraft S/N 5022 completion and cabin interior installation.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC52 0001129, Revision J, dated 16 June2004, or later approved revisions Operations to be in accordance with Gulfstream Flight Manual Supplement CE 10058600, original release dated 28 June 2007, or later approved revisions...</td>
<td>GV-5P S/N 5022 only.</td>
<td>26/07/2007</td>
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<td>EASA.IM.A.S.02450</td>
<td>UNIVERSAL AVIONICS SYSTEMSCORPORATION</td>
<td>FAA STC ST01879LA</td>
<td>BOMBARDIER CL-600-2B16 (601-3A)</td>
<td>Universal Avionics Vision-1™ Synthetic Flight Guidance System installed in Canadair Challenger CL-600 Series.</td>
<td>Universal Avionics Master Drawing List MDL-2005-1003 Revision C dated 23 May 2006 or later approved revision Universal Avionics Airplane Flight Manual Supplement AFMS-2005-1003 Revision II dated 17 May 2006 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationships between the modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/11/2007</td>
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<td>EASA.IM.A.S.02452</td>
<td>ODGERICH CORPORATION</td>
<td>FAA STC ST02581CH</td>
<td>BOEING 747-400F</td>
<td>Installation of Centerline Restraint System on the main cargo deck of a B747-400F in accordance with Master Data List MDL 7439000. Validation of FAA STC ST02581CH.</td>
<td>MD0522 Aircraft weight and balance control and loading manual supplement MD0521 Aircraft cargo loading manual supplement MD0535 Aircraft maintenance manual supplement.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/07/2008</td>
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<td>EASA.IM.A.S.02454</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA SA03-113</td>
<td>CL-600-2B16</td>
<td>Complete Custom Aircraft Interior.</td>
<td>TCCA STC SA03-113.Aero Consulting Services Configuration Definition List C817150 Revision P, dated 4-07-2007 Aero Consulting Services AFM D817150 Revision N/C or later approved revisions.</td>
<td>Restricted to ASN 5628.</td>
<td>06/06/2007</td>
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<td>EASA.IM.A.S.02455</td>
<td>LANDMARK AVIATION</td>
<td>FAA STC ST01136CH-D</td>
<td>HS125 SE.600A, HS125 SE.700A, HS126 DH125 SE.400A, DH125 SE.3A</td>
<td>Installation of a Senior Flexonics Inc. Signal Conditioner Unit</td>
<td>Installation of Senior Flexonics Inc. Signal Conditioner Unit.</td>
<td>Compatibility of this design change with previously approved modifications must be determined by the installer. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person evidence of that permission.</td>
<td>31/05/2007</td>
<td>Active</td>
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### List of EASA Supplemental Type Certificates

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<th>EASA STC No.</th>
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<tr>
<td>EASA.IM.A.S.02456</td>
<td>AERO MEDICAL PRODUCTS MFG. Inc.</td>
<td>FAA STC SA105QL</td>
<td>PA-31, PA-31-300, PA-31-325, PA-31-350 (CHEFTRIM)</td>
<td>FAA STC SA105QL, “Armair Stretchy System - Installation/ Air Ambulance Stretchy PIN 125 KI PIN 800, Medical Oxygen System Kt PIN 807 (FAA approved option). Invertair Pump/PIN 812 in accordance with Drawing List Report 001 Rev.C or later approved revision.</td>
<td>- AFM Supplement EA Form FMS FAA Approved 09-02-2004 or later approved revision - AMM Supplement EA Forms MMS FAA Approved 10-02-2004 or later approved revision - Installation and Operating Instructions EA Form 108P FAA approved 23-10-2001 or later approved revision - Maintenance Requirements.</td>
<td>1. Prior to the first flight of each modified aircraft, and after each exchange to any of the Communication, Navigational or Meteorological Equipment, a ground EMI test must be conducted in accordance with EA Form 173 dated 15-11-1993. A copy of the EA Form 173 must be kept with the aircraft records.</td>
<td>01/02/2008</td>
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<td>EASA.IM.A.S.02460</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST01119W-D</td>
<td>MYSTERE FALCON 50, HAWKER HS 125-700B, GULFSTREAM IV, GULFSTREAM 1159 (G-II), GULFSTREAM 1159 B (G-II), FALCON 900 EX, FALCON 10, CESSNA 700</td>
<td>Modification of Collins TDR-64 or TDR-44D Mode S Transponder System installation with stand-alone Collins TRU-4210 or RTU-4200 Radio Tuning Unit(s) to a Flight ID capable Collins TDR-64 or TDR-44D Mode S Transponder System installation with stand-alone Collins RTU-4210 or RTU-4220.</td>
<td>In accordance with Collins Installation Instructions as listed on AML No. ST01119W-D. Revision-, or later approved revision.</td>
<td>(1) Approval of this change in type design applies to the above model aircraft only (2) Compatibility of this design change with previously approved modifications must be determined by the installer. (3) A copy of this Certificate and FAA Approved Model List (AML) No. ST01119W-D issued October 29, 2007.</td>
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<td>EASA.IM.A.S.02462</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC C-LSA02-251D</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Customized Interior Completion - SN 9084/Validation of TCCA STC C-LSA02-251D issue 2</td>
<td>AFM Supplemen...</td>
<td>Refer to the Limitations sections of the required AFM Supplement</td>
<td>05/10/2007</td>
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<td>EASA.IM.A.S.02464</td>
<td>DASSAULT FALCON JET CORP (SERVICE CENTER)</td>
<td>FAA STC SAB312SW-D</td>
<td>FALCON 50</td>
<td>Installation of the Automatic Cabin Electrical Load Shed System in accordance with Drawing List Report No. FS-01182 Revision A, dated 27th September 1999 or later approved revisions.</td>
<td>- FAA STC No. SAB312SW-D - FAA Approved Airplane Flight Manual Supplement Report No. 39627-1 dated 19th February 1998 or later approved revision is required for the -501 installation - FAA Approved Airplane Flight Manual Supplement Report No. 39627-2 dated 30th September 1998 or later approved revision - FAA approved 23-10-2001 or later approved revision.</td>
<td>1. As per FAA STC No. SAB312SW-D. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installer...</td>
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<td>DASSAULT FALCON JET CORP (SERVICE CENTER)</td>
<td>FAA STC SA8295SW-D</td>
<td>FALCON 50</td>
<td>Installation of the Auxiliary Battery System in accordance with Drawing List Report No. F50-01173, Revision E, dated 19th September 2001 or later approved revisions.</td>
<td>FAA STC No. SA8295SW-D • FAA Approved Airplane Flight Manual Supplement Report No. 36587-1 dated 3rd June 1997 or later approved revision is required for the -501 installation of the Drawing List • FAA Approved Airplane Flight Manual Supplement Report No. 36587-2 dated 18th August 2000 or later approved revisions.</td>
<td>1. As per FAA STC No. SA8295SW-D.2 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installer...</td>
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<td>EASA.IM.A.S.02466</td>
<td>DASSAULT FALCON JET CORP (SERVICE CENTER)</td>
<td>FAA STC SA8222SW-D</td>
<td>FALCON 50</td>
<td>Installation of the GWS Satellite Data Communication System in accordance with Drawing List Report No. F50-01159, Revision NC, dated 21st July 1993 or later approved revisions.</td>
<td>FAA STC No. SA8222SW-D.</td>
<td>1. As per FAA STC No. SA8222SW-D.2 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installer...</td>
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<td>EASA.IM.A.S.02467</td>
<td>DASSAULT FALCON JET CORP (SERVICE CENTER)</td>
<td>FAA STC SA7300SW-D</td>
<td>FALCON 50</td>
<td>Installation of the Service and Indirect Lighting System in accordance with Drawing List Report No. F50-01416, Revision LC, dated 25th October 2000 or later approved revisions.</td>
<td>FAA STC No. SA7300SW-D.</td>
<td>1. As per FAA STC No. SA7300SW-D.2 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installer...</td>
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<td>DASSAULT FALCON JET CORP (SERVICE CENTER)</td>
<td>FAA STC ST09316SC-D</td>
<td>FALCON 50</td>
<td>Installation of the Standby Instrument System in accordance with Drawing List Report No. F50-01172, Revision B, dated 12th October 2000 or later approved revisions.</td>
<td>FAA STC No. ST09316SC-D.2 FAA Approved Airplane Flight Manual Supplement Report No. 36586; Revision A, dated 26th May 2000 or later approved revision is required when -501 or -503 installations are used.</td>
<td>1. FAA STC No. ST09316SC-D.2 Limited to Falcon 50EX version aircraft.3 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installer...</td>
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<td>EASA.IM.A.S.02469</td>
<td>DASSAULT FALCON JET CORP (SERVICE CENTER)</td>
<td>FAA STC SA7260SW-D</td>
<td>FALCON 50</td>
<td>Installation of the Auxiliary Equipment Rack in accordance with Drawing List Report No. F50-01116, Revision N, dated 2nd June 1993 or later approved revisions.</td>
<td>FAA STC No. SA7260SW-D.</td>
<td>1. As per FAA STC No. SA7260SW-D.2 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installer...</td>
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<td>DASSAULT FALCON JET CORP (SERVICE CENTER)</td>
<td>FAA STC SA7068SW-D</td>
<td>FALCON 50</td>
<td>Installation of the Centralized Water System in accordance with Drawing List Report No. F50-01005, Revision A, dated 6th September 1999 or later approved revisions.</td>
<td>FAA STC No. SA7068SW-D. • Airplane Maintenance Manual Supplement Report 35669 Revision A or later approved revisions is applicable to those aircraft modified i.e. with this STC and having the heated belly drain modification installed as per drawing F50-41813.</td>
<td>1. As per FAA STC No. SA7068SW-D.2 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installer...</td>
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<td>EASA.IM.A.S.02472</td>
<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>FAA STC ST10423SC-D</td>
<td>FALCON 900EX</td>
<td>Installation of the Service and Indirect Lighting System (EMTEG) in accordance with Data List Report No. F900EX-01048, Revision A dated 07th October 2005 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or i.a.w. subsequent revisions of th...</td>
<td>FAA STC ST10423SC-D</td>
<td>1. As per FAA STC ST10423SC-D.2. Applicable to Falcon900EX EASA aircraft version 3. F900EX EASA designation does not correspond to a new MF900 model designation but is a commercial designation for airplanes on which the specific Dassault modifications as listed on EASA TCDs EASA.A.082 have...</td>
<td>26/02/2007</td>
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<td>EASA.IM.A.S.02476</td>
<td>FIELD AVIATION COMPANY INC.</td>
<td>TCCA C-LSA07-021/D</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Installation of Corporate Shuttle Interior.</td>
<td>STC C-LSA07-021/D, Issue 1 dated 16 September 2007. Installation of the corporate shuttle interior must be in accordance with Transport Canada Civil Aviation (TCCA), approved Flight Operation Company Inc. (FAC), Modification instruction 209410, Revision A, dated August 31, 2007, or later TCCA approved.</td>
<td>Limited to serial no. 8049</td>
<td>09/10/2007</td>
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<td>EASA.IM.A.S.02478</td>
<td>LEARJET Inc.</td>
<td>FAA STC SA1811NM-D</td>
<td>BOMBARDIER CL-600-2B16 (CL-604</td>
<td>FAA STC SA0859WI-D, NAV Lights Relocation/This STC supercedes LBA TAI136.</td>
<td>Bombardier Aircraft Services Mix No. CA-3017-TU at Revision F dated 06 June 2002 or later approved revision</td>
<td>31/05/2007</td>
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<td>EASA.IM.A.S.02480</td>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD</td>
<td>TCCA STC Q-LSA07-029/D</td>
<td>BD-700-1A11</td>
<td>Installation of Photosmart printer and installation of additional handset for in the aft lhs of SN 9204/Validation of TCCA Limited STC Q-LSA07-029/D in SN 9204 Aircraft.</td>
<td>Compliance plan: CP27116-601.</td>
<td>21/01/2006</td>
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<td>GULFSTREAM AEROSPACE CORPORATION.</td>
<td>FAA STC ST10314SC-D</td>
<td>GULFSTREAM G500</td>
<td>Installation of a Satcom Antenna.</td>
<td>IFGA GA34103A000</td>
<td>09/10/2007</td>
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<td>EASA.IM.A.S.02492</td>
<td>HAWKER BEECHCRAFT CORPORATION (HBC)</td>
<td>FAA STC SA0906W</td>
<td>BEECH 1900D</td>
<td>Installation of EFOS MAGIC (Mask and Goggles Integrated Concept) Quick Donning Masks.</td>
<td>Installation in accordance with Raytheon Aircraft Services Master Drawing List 148-100057, Revision A, dated January 27, 2000, or later approved revision.</td>
<td>17/07/2007</td>
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<td>EASA.IM.A.S.02496</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST02848AT-D</td>
<td>GIV, GV</td>
<td>Installation of the Runway Awareness and Advisory System (RAAS) Software for the Honeywell MK V Enhanced Ground Proximity Warning System (EGPWS)</td>
<td>Installation in accordance with Gulfstream Index List CE413498050, revision F dated February 21, 2008 or later FAA approved revision. Operation to be in accordance with Gulfstream Flight Manual Supplements CE413498008. Original Release, dated February 21, 2008, later approved revisions for the mod...</td>
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<td>EASA.IM.A.S.02497</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0428AT-D</td>
<td>STC Holder</td>
<td>GULFSTREAM GV-SP</td>
<td>Cabin Interior and Aircraft Completion STC on Aircraft Serial Number 5150.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC51 491515, Rev. C, dated August 21, 2007, for later approved revision. To be operated in accordance with FAA approved Airplane Flight Manual Supplement ref. GC511501022, Rev. NC, dated 28/07/07, or later approved revision.</td>
<td>Aircraft Serial Number 5150 only. The following STCs are a required part of this STC: ST02694AT-D, ST02697AT-D and ST03290AT-D. It is the installer’s responsibility to ensure these STCs are all EASA approved. Prior to installation of this modification the installer must determine it...</td>
<td>10/02/2007</td>
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<td>EASA.IM.A.S.02498</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>STC ST03414AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5153.</td>
<td>Installation in accordance Gulfstream Aerospace Corporation Master Index List, GC514915153 Rev. B, dated September 4, 2007, or later approved revisions.</td>
<td>Aircraft Serial Number 5153 only. The following STCs are a required part of this STC: ST02694AT-D, ST02697AT-D, ST03290AT-D, ST02715AT-D, ST02796AT-D and ST00892LA. It is the installer’s responsibility to ensure these STCs are all EASA approved.</td>
<td>09/10/2007</td>
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<td>EASA.IM.A.S.02499</td>
<td>DUNCAN AVIATION, INC.</td>
<td>FAA STC ST0686WI</td>
<td>LEARJET 36A LEARJET 36 LEARJET 35A LEARJET 35</td>
<td>Installation of Collins TCAS-94 TCAS II System.</td>
<td>AFM Supplement 40003500, ref.</td>
<td>This installation is not to be used in airspace where Elementary Surveillance (ELS) or enhanced surveillance (EHS) is required.</td>
<td>19/12/2007</td>
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<td>EASA.IM.A.S.02500</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST01937LA-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5151.</td>
<td>Install in accordance with Gulfstream Aerospace Report GC5000K477, Rev. N/C and Index List GC520008576 Rev. K, dated 06 September 07, or later approved revisions. Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M045, Revision N/C, dated 06 September 07, or later approved revisions.</td>
<td>Aircraft Serial Number 5151 only.</td>
<td>09/10/2007</td>
<td>Active</td>
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<td>EASA.IM.A.S.02501</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA C-LSA07-366/D</td>
<td>BD-100-1A10</td>
<td>AIRCRAFT COMPLETION: 10 PASSENGERS INTERIOR CONFIGURATION: EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA07-366/DIssue 1 dated 20th December 2007, in accordance with Aero Consulting Services Ltd. Modification Data Summary FS76000 Revision N/C.</td>
<td>Transport Canada Limited Supplemental Type Certificate C-LSA07-366/D Issue 1, dated 20th December 2007. Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary ref. FS76000 dated 20th December 2007, Revision NC or later Transport Canada approved revision.</td>
<td>This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20169.</td>
<td>10/01/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.02502</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA C-LSA07-320/D</td>
<td>BD-100-1A10</td>
<td>AIRCRAFT COMPLETION: 10 PASSENGERS INTERIOR CONFIGURATION: EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA07-320/DIssue 1 dated 20th November 2007 in accordance with Aero Consulting Services Ltd. Modification Data Summary FS72200 Revision N/C dated 19.11.07</td>
<td>Transport Canada Limited Supplemental Type Certificate C-LSA07-320/D Issue 1, dated 20th November 2007. Transport Canada Approved Aero Consulting Services Ltd. Airplane Flight Manual Supplement FS72200 Revision N/C or later Transport Canada approved revision.</td>
<td>This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20159.</td>
<td>26/11/2007</td>
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<td>EASA/IM.A.S.02512</td>
<td>BUTLER NATIONAL CORPORATION</td>
<td>FAA STC NO. ST006460E</td>
<td>BOEING 747-300 BOEING 747-200 BOEING 747-100</td>
<td>Installation of Model 84037-108, 84037-112, and 84037-113 transient Suppression Device on Boeing 747-100, 747-200, 747-300 SP and SR series airplanes fitted with Honeywell Analog Fuel quantity Indication System in accordance with ButlerNational Corporation document No. TSD BN 0031, Drawing Packa...</td>
<td>TCCA STC</td>
<td>10016247</td>
<td>This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other acroaendance configurations shall be determined by the installer.</td>
<td>15/10/2007</td>
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<td>EASA/IM.A.S.02515</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TOCA STC SA55-40</td>
<td>TOCA STC SA55-40</td>
<td>Customized Interior Completion - SN 9182Validation of TOCA STC SA55-60 via SoC. Limited to SN 9182-6 and installation acc. CDL E25000 Rev N</td>
<td>FAA STC</td>
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<td>Refer to the Limitations sections of the required TCCA STC Supplement.</td>
<td>24/10/2007</td>
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<td>EASA/IM.A.S.02517</td>
<td>ROCKWELL COLLINS Inc.</td>
<td>FAA STC ST0056LA</td>
<td>AIRBUS A330-203 AIRBUS A330-203</td>
<td>IFE VMOD upgrade per System List 1310629-001IFE VMOD Upgrade consists of removing the existing VMOD and re-installing an upgraded VMOD in the Video Control Center (VCC).</td>
<td>FAA STC</td>
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<td>EASA/IM.A.S.02518</td>
<td>ROCKWELL COLLINS Inc.</td>
<td>FAA STC ST0056LA</td>
<td>AIRBUS A340-313 AIRBUS A340-311</td>
<td>IFE VMOD upgrade per System List 1310630-001IFE VMOD Upgrade consists of removing the existing VMOD and re-installing an upgraded VMOD in the Video Control Center (VCC).</td>
<td>FAA STC</td>
<td></td>
<td>Prior to installation of this modification the installer must determine that the installation does not affect the interrelationship between this modification and any other previously installed modification.</td>
<td>04/10/2007</td>
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<td>EASA/IM.A.S.02520</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>GULFSTREAM GIV-X</td>
<td>Aircraft Serial No 4092 completion and cabin interior installation.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC422046L19, Revision B, Dated November 16, 2007, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC422046M33, Original release, dated November 20, 2007, or later approv...</td>
<td>FAA STC ST0056LA-001</td>
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<td>GIV-X Serial No. 4092 Only.</td>
<td>10/01/2008</td>
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<td>EASA/IM.A.S.02524</td>
<td>HAWKER BEECHCRAFT SERVICES</td>
<td>FAA STC ST02100AT</td>
<td>BEECH 400</td>
<td>Installation of a Battery Low Voltage Warning / Shutoff System on Raytheon Aircraft 400 and 405A.</td>
<td>FAA Master Data List</td>
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<td>EASA/IM.A.S.02525</td>
<td>HAWKER BEECHCRAFT SERVICES</td>
<td>FAA STC SA03200AT</td>
<td>BEECH 390</td>
<td>Modification to install an Auxiliary Ground Heating System.</td>
<td>FAA STC ST02100AT</td>
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<td>HAWKER BEECHCRAFT SERVICES</td>
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<td>HAWKER 800 XP HAWKER 800</td>
<td>Hawker Beechcraft Corporation Installation of upper and lower LED beacons, wingtip LED navigation lights and wingtip strobe lights.</td>
<td>Installation as defined in Hawker Beechcraft Master Data List RAS050600152 rev 3.0, dated May 11, 2007, or later approved revisions. Instructions for Continued Airworthiness as defined in - Hawker Beechcraft (ICA) document ref RAS050600161, latest approved revision.</td>
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<td>EASA.IM.A.S.02529</td>
<td>ELLIOTT AVIATION</td>
<td>FAA STC SAU4447CH</td>
<td>BEECH 300C BEECH 300B BEECH 2007 BEECH 200CT BEECH 200C BEECH 200i</td>
<td>Installation of a Rockwell Collins IFIS-5000 Integrated Flight Information system.</td>
<td>Installation must be in accordance with Elliott Aviation Master Data List No. MDA-2707-000 Rev. A, dated 26. June 07 or later approved Revisions. Operation must be in accordance with Elliott Aviation Flight Manual Supplement FMS-2707-00 Rev. IR dated 05 July 07 or later approved revision. Elliott Aviation Flight Manual Supplement ref. FMS-2707-00 Rev. IR dated 05 July 07 or later approved revision. Maintenance m...</td>
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<td>EASA.IM.A.S.02531</td>
<td>UNIVERSAL AVIONICS SYSTEMS CORPORATION</td>
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<td>Universal Avionics UNS-1Fw with embedded WAAS: MDA-4 Config (TripleUNS-1Fw FMS with P/N 10709 3 GPS-WAAS/Sirius Antenna)</td>
<td>EASA.IM.A.023 - EASA STC No. EASA.IM.A.023 - FAA STC No. ST02070LA Universal Avionics MDA No. MDA-2007-1003 Revision 11 dated 11 January 2009 or later approved revision. Universal Avionics MDA Supplement No. AFMS-2007-1006 Revision 00 dated 31 January 2008 or later approved revision</td>
<td>None</td>
<td>06/07/2009</td>
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## List of EASA Supplemental Type Certificates

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<td>EASA.IM.A.S.02532</td>
<td>ELLIOTT AVIATION</td>
<td>FAA STC ST02431CH</td>
<td>FALCON 10</td>
<td>ESAVA Validation of FAA STC ST02431CH.ESAVA Validation is limited to Provisions for Installation of a Universal Avionics Application Server Unit (ASU) on Falcon 10</td>
<td>Installation in accordance with: - Elliott Aviation Master Data List, Report No. MDL-271-00. Revision A, dated 15th may 2007, and with - Elliott Aviation drawing no. 300-2710-EOA0300-001 Rev IR dated 06th May 2008 for the deactivation of the ASU System and for the improvements for installation. Spec...</td>
<td>The aircraft must be equipped with Universal Avionics EISF-860R Electronic Flight Instrument System (EFIS) as installed by Elliott Aviation’s STC EASA.IM.A.S.01713. This EASA Approval is limited to provisions for installation only with the ASU System deactivated. A placard indicating the ASU System is...</td>
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<td>EASA.IM.A.S.02533</td>
<td>SAFRAN CABIN BELLINGHAM, INC.</td>
<td>FAA STC ST01790SE</td>
<td>B747-400</td>
<td>Installation of interior components in accordance with Health Tecna Inc. Drawing List HPD-DL-105</td>
<td>Health Tecna Inc. Drawing List HPD-DL-105. Revision B, dated July 5, 2007 FAA STC ST01790SE or later revisions of the above listed documents approved by EASA in accordance with ED Decision 2004/49/CF (or subsequent revisions of this decision). MODIFICATION of the aircraft per Boeing Service Bulletin 747-00-2004, conversion from passenger to freighter configuration must be performed concurrently with this STC.</td>
<td>Modification of the aircraft per Boeing Service Bulletin 747-00-2004, conversion from passenger to freighter configuration must be performed concurrently with this STC.</td>
<td>29/08/2007</td>
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<td>EASA.IM.A.S.02534</td>
<td>SAFRAN CABIN BELLINGHAM, INC.</td>
<td>FAA STC ST06700SE</td>
<td>B737-330</td>
<td>Installation of galley supports structure and jumper wire provisions in accordance with Health Tecna Drawing List HPD-DL-76. Revision B, dated 16 January 2001 or later approved revision.</td>
<td>Health Tecna Drawing List HPD-DL-76. Revision B, dated 16 January 2001 or later approved revision.</td>
<td>Approval of this change in type design applies to the above model aircraft only. This approval should not be extended to aircraft on which other previously approved modifications are incorporated unless it is determined that the relationship between this change and any of those other previously approved changes and any of those other previously approved changes (or subsequent revisions of the above listed documents approved by EASA in accordance with ED Decision 2004/49/CF (or subsequent revisions of this decision). Mod...</td>
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<td>EASA.IM.A.S.02535</td>
<td>AS&amp;T AEROSPACE SYSTEM &amp; TECHNOLOGIE INC</td>
<td>FAA SA01328WI</td>
<td>2-PW32R-301T (TURBO SARRATOGA)  2 - PA-32R-301 (SARRATOGA SP)  1 - PA-32-301XT (PIPER 6X)  1 - PA-32-301FT (PIPER 6X)</td>
<td>Installation of TKS Ice Protection System Components on no-hazard / no credit basis.</td>
<td>Installation must be in accordance with AS&amp;T Master Drawing List No. 12700DL. Revision 3, dated March 7, 2005 or later approved revision. Operation must be in accordance with Supplemental Airline Flight Manual and FAA approved Airline Flight Manual Supplement, Doc. No. 12700-FMS, dated 13 May 2005...</td>
<td>1. This installation is not approved for flight into known icing conditions 2. A placard must be fitted on the upper control panel in front of the pilot, stating &quot;FLIGHT INTO KNOWN ICING CONDITIONS IS PROHIBITED. This installation is approved on a no-hazard / no credit basis.</td>
<td>27/06/2007</td>
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<td>EASA.IM.A.S.02540</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>FAA STC ST4837WI-T</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Installation of an Audio International Articulated Mounting Arm and Data Connectivity for an Electronic Flight Bag</td>
<td>FAA STC No. ST02544CH-D dated 07 March 2008, Aviation, Inc. Master Data List 41-3432001 Revision C, dated March 07, 2008, or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>03/10/2008</td>
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<td>10016261</td>
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<td>EASA.IM.A.S.02541</td>
<td>HOLLINGSEAD INTERNATIONAL</td>
<td>FAA STC ST04046LA-D</td>
<td>BOEING 737-400</td>
<td>Installation of Smoke Detection and Fire Suppression Systems Provisions, Lower Cargo Compartments.</td>
<td>Drawing No 90437 (-506&amp;-507) Rev C dated 03/23/99, and MDL 12111 Rev. D dated 10/15/99, and MDL 12111 Rev. D dated 05/02/00FAA approved Aircraft Flight Manual Supplement 20673 Rev.B dated 02/28/00 or later approved revisions of the above list...</td>
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<td>14/01/2008</td>
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<td>EASA.IM.A.S.02544</td>
<td>GULFSTREAM AEROSPACE CORPORATION.</td>
<td>FAA STC ST10317SC-D</td>
<td>GULFSTREAM G150</td>
<td>Installation of a Collins IFIS 5000 EFB System.</td>
<td>AFMS: GA31120M000 or GA31120M001IFCA: GA31120A000</td>
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<td>10/10/2007</td>
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<td>EASA.IM.A.S.02545</td>
<td>GULFSTREAM AEROSPACE CORPORATION.</td>
<td>FAA STC ST10321SC-D</td>
<td>GULFSTREAM G150</td>
<td>Installation of a WX-1000 Stormscope</td>
<td>AFMS: GA31250M000IFCA: GA31250A000</td>
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<td>09/10/2007</td>
<td>Active</td>
<td>10016265</td>
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<td>EASA.IM.A.S.02547</td>
<td>GULFSTREAM AEROSPACE CORPORATION.</td>
<td>FAA STC ST10327SC-D</td>
<td>GULFSTREAM G150</td>
<td>Installation of a Colntes DBU 5000 System.</td>
<td>AFMS: GA31250M000IFCA: GA31250A000</td>
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<td>09/10/2007</td>
<td>Active</td>
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<td>EASA.IM.A.S.02548</td>
<td>GULFSTREAM AEROSPACE CORPORATION.</td>
<td>FAA STC ST10328SC-D</td>
<td>GULFSTREAM G150</td>
<td>Installation of a second ADI.</td>
<td>AFMS: GA31250M000IFCA: GA31250A000</td>
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<td>05/12/2007</td>
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<td>EASA.IM.A.S.02549</td>
<td>GULFSTREAM AEROSPACE CORPORATION.</td>
<td>FAA STC ST103286SC-D</td>
<td>GULFSTREAM G150</td>
<td>Installation of a Honeywell FDR.</td>
<td>AFMS: GA31420M000IFCA: GA31420A000</td>
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<td>09/10/2007</td>
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<td>EASA.IM.A.S.02552</td>
<td>HAWKER BEECHCRAFT SERVICES</td>
<td>FAA STC ST04040AT</td>
<td>BEECH 400A BEECH 400</td>
<td>Replacement of a Standby Attimeter installation in the Raytheon (Beech) models 400, 400A aircraft. This STC is a duplicate of STC ST0379AT. EASA IM AS 02353</td>
<td>AFMS: GA31214M000 IFCA: GA31214A000</td>
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<td>04/06/2007</td>
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<td>EASA.IM.A.S.02554</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA C-LSA07-165D ISSUE 2</td>
<td>767-300</td>
<td>Upgrade to Honeywell Mark V EGPHS.</td>
<td>Aero Consulting Services Modification Data Summary F79000 rev B Flight Manual Supplement, F79091, rev N C Instructions for Continued Airworthiness, F79101, rev N C</td>
<td>Prior to installation of this modification the installer must determine the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
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<td>Date of Issue</td>
<td>Aircraft System</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>EASA.IM.A.S.02567</td>
<td>COLUMBIA AVIONICS INC.</td>
<td>FAA STC ST01252WI</td>
<td>Active</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>21/06/2007</td>
<td>10016272</td>
<td>C-L000-1A11 (C-L000 Series)</td>
<td>Vertical Separation Minimum (VSM) group airworthiness approval of Cessna Citation aircraft</td>
<td>Master Doc. 231245 Rev. 3 of June 4, 2003 instructions for initial and continued airworthiness, Doc. 3014236 Rev. A of April 23, 2004 or later FAA approved revisions to these two documents.</td>
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<td>EASA.IM.A.S.02568</td>
<td>ROCKWELL COLLINS, Inc. of Air Transport Systems</td>
<td>FAA STC ST00392AT</td>
<td>Active</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>02/11/2007</td>
<td>10016273</td>
<td>TC STC C-LS07-265D ISSUE 2</td>
<td>Installation of Collins Dual TPR-720 Mode S Transponders</td>
<td>Collins Master Drawing List, MCL-95-38, rev. F, 22/03/2000.</td>
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<td>EASA.IM.A.S.02569</td>
<td>CORPORATION LOGIC AIR Inc.</td>
<td>REV 1 TC STC C-LS07-265D ISSUE 2</td>
<td>Active</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>10/04/2008</td>
<td>10016274</td>
<td>BOEING 767-300</td>
<td>Installation of In-Flight Entertainment System</td>
<td>1. Aero Consulting Service Modification Data Summary F37100, rev. A. 2. Instruction for Continued Airworthiness F37107, rev. NC for S/N 27957 and 26263; F37107, rev. NC for S/N 27135 and 37140</td>
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<td>EASA.IM.A.S.02572</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC ST01470LA-D</td>
<td>Active</td>
<td>None.</td>
<td>08/07/2006</td>
<td>10016275</td>
<td>737-800, 737-900</td>
<td>Installation of Rockwell Collins VR-2100 Multi-scan (TM) Weather RadarSystemRevision to the existing STC EASA.IM.A.S.02562 to change the STC holder's name and to add more models and correct the equipment reference name</td>
<td>Hollingshead International Inc. Master Data List No 2736-12 Rev A dated 28-03-04, or later approved revision.</td>
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<td>EASA.IM.A.S.02573</td>
<td>AIRCRAFT SYSTEMS AND MANUFACTURING G, INC.</td>
<td>FAA STC ST10288BC</td>
<td>Active</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>10/10/2007</td>
<td>10016276</td>
<td>DC-8-71/-72/-73/-7SF DC-8-61/-61F/-62/-62F/-63/-63F</td>
<td>FAA STC ST10288BC &quot;Installation of Universal Avionics Terrain Awareness Warning System (TAWS).&quot;</td>
<td>Aircraft Systems &amp; Manufacturing Top drawing List 5943-0001-00A dated 20 July 2004 AFM Supplement No. 5843-2100-00 Rev. IR dated 22 July 2004.</td>
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<td>EASA.IM.A.S.02574</td>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD.</td>
<td>Q-LS07-035/ D</td>
<td>Active</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>12/10/2007</td>
<td>10016277</td>
<td>BD-700-1A11</td>
<td>Miscellaneous Cabin Modifications on SN 9188 Airworthiness Approval of Collins Master Drawing List 2003-1240-00 Compliance Plan. CP27124-001 Rev. A</td>
<td>IFCA TL27124-004Soc. NIA5012-07-0128, dated August 28, 2008</td>
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<td>STC Holder</td>
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<td>Models</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST02390NY-D</td>
<td>BOEING 747-400</td>
<td>Installation of Heath Tecna Cabin Interior</td>
<td>Master data list 0107-04068-2152, Rev A, and instructions for Continued Airworthiness 0707-04050-2152, Rev A, or later approved revisions.</td>
<td>1. TGL 12 complied with. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>09/10/2007</td>
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<td>HAWKER BEECHCRAFT SERVICES</td>
<td>FAA STC/STC ref</td>
<td>BEECH 300</td>
<td>Installation of a 70% Ground Idle Stop Dropout System Installs cams and switches on the existing aircraft’s power lever system to allow the propeller system to transition from Ground Idle to Flight Idle when both Power levers are advanced above 70 %. This allows the propellers to attain the proper...</td>
<td>Installation in accordance with Raytheon Aircraft Services Master Drawing List Number MDL-TAF96106, Original Revision, dated February 12, 1997, or later FAA approved revision. Operation in accordance with Raytheon Aircraft Services Aircraft Flight Manual Supplement, dated May 29, 1997 Maintenance instructions.</td>
<td>Aircraft Serial Number 4093 only. The following STCs are a required part of this STC:ST03290AT-DST03047AT-DST02980AT-DST02979AT-DST02997AT-DST03295AT-DST02987AT-DST03047AT-DST02980AT-DST02979AT-DST02997AT-DST03295AT-DST02987AT-DST03047AT</td>
<td>19/10/2007</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC/STC ref</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4093.</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC4914050 Rev. D, dated November 26, 2007, or later approved revisions.</td>
<td>Aircraft Serial Number 4093 only. The following STCs are a required part of this STC:ST03290AT-DST03047AT-DST02980AT-DST02979AT-DST02997AT-DST03295AT-DST02987AT-DST03047AT-DST02980AT-DST02979AT-DST02997AT-DST03295AT-DST02987AT-DST03047AT</td>
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<td>CORPORATION LOGIC AIR Inc.</td>
<td>FAA STC/STC ref</td>
<td>BOEING 767-300</td>
<td>Tire pressure Indication System De-activation; Boeing 767-300.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>07/06/2008</td>
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<td>CORPORATION LOGIC AIR inc.</td>
<td>FAA STC/STC ref</td>
<td>BOEING 767-300</td>
<td>Installation of Mode S ATC Transponder with Elementary and Enhanced Surveillance Modes.</td>
<td>Aero Consulting Services Modification Data Summary F276080 rev NC, Flight Manual Supplement E276062, rev N, C, Instruction for Continued Airworthiness Supplement E276071, rev N, C,</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>FAA STC/STC ref</td>
<td>VIKING AIR LIMITED (DCH) DHC-6</td>
<td>Dual or Single Garmin GNS550 or GNS430 installation.</td>
<td>Installation in accordance with Field Aviation Company, Inc. Modification Instruction 181811, original Issue or later approved revision for single GNS-340 or GNS-530 Installation; 181813, original Issue or later approved revision for singleGNS-430W or GNS-530W Installation; 181810, Rev. B or later a...</td>
<td>Prior to installation of this modification the installer must determine the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
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<td>EASA.IMA.S.02562</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>REV. 1</td>
<td>FAA STC ST01927/7A</td>
<td>BOEING 757-300 BOEING 757-200</td>
<td>Installation of ARINC Characteristics 718A for Mode S Transponder Activation of Enhanced Surveillance Capabilities.</td>
<td>• Hollis Engineering International Master Data List 2894-01, Revision NC, 19th October 2006. • AFM Supplement 2894-AFM, Revision Original, 24th October 2006.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>15/10/2007</td>
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<td>EASA.IMA.S.02563</td>
<td>LIFEPORT, LLC</td>
<td>FAA STC SA02060/6LA</td>
<td>BEECH 1900D</td>
<td>Emergency Medical System</td>
<td>FAA approved Master Drawing List, Doc No. MSL-822-00, Rev E, dated 09-Jul-2007, or later approved revision/FAA approved Airplane Flight Manual Supplement, Doc. No. EGR-822-11, Rev. A, dated 31-Jul-2007, or later approved revision/instructions for Continued Airworthiness, Doc. No. EGR-822-12, Rev. A.</td>
<td>The installer of this STC must ensure that, if the life vests have been removed with the seats, a life vest for the occupant of the heater system is installed and available to the occupant.</td>
<td>04/10/2007</td>
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<td>EASA.IMA.S.02569</td>
<td>BOMBARDIER INC.</td>
<td>TCCA STC SA06-84</td>
<td>BOMBARDIER BD-100-1A10 (CHALLE)</td>
<td>Improved Hinge Design, Toilet Seat and Lid / Modification of existing toilet seat installation which allows both toilets and lid to remain in the up position without failing EASA Validation of Transport Canada Supplemental Type Certificate SA06-84, Issue 1 dated 3rd November 2006 in accordance w...</td>
<td>-Transport Canada Supplemental Type Certificate SA06-84 Issue 1, dated 3rd November 2006. -Aero Consulting Modification Data Summary, ref. No. F02000 Revision A, dated 02/08/2007. -Decree Aircraft Service Bulletin &quot;Bombardier C300 Laviatory Seat Lid Hinge Assembly&quot;, ref. SB06008, Revision G, dated 23rd July 2007.</td>
<td>05/11/2007</td>
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<td>EASA.IMA.S.02591</td>
<td>NORTHEAST ENGINEERING &amp; DEVELOPMENT, TD</td>
<td>TCCA STC NO. G-LSA05-321/D IS</td>
<td>PILATUS PC-12/45</td>
<td>Installation of GNS 430 (COM/MNAV/GPS), GNS 530 (COM/MNAV/GPS), HDR 510 (Data link Weather Receiver), KG5 SSSI (Static Inverter), XM Radio Receiver with XM Radio Display &amp; Control Head (Cabin), and necessary antennas.</td>
<td>Installation of GNS 430 (COM/MNAV/GPS), GNS 530 (COM/MNAV/GPS), HDR 510 (Data link Weather Receiver), KG5 SSSI (Static Inverter), XM Radio Receiver with XM Radio Display &amp; Control Head (Cabin), and necessary antennas.</td>
<td>This STC is limited to Pilatus PC-12/45, Serial number 6589/Prior to the installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the pr...</td>
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<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>Model(s)</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>Associated Technical Documentation</td>
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<td>Model(s)</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>TCCA STC C-LSA07-286/D FOR CAB</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Customised Interior Completion - ACSN 9238.</td>
<td>TCCA SOC letter C-07-0767 dated 17 January 2008. Aero Consulting Services Modification Data Summary F482000 Revision N/C dated 10 December 2007. Bombardier Airplane Flight Manual Supplement G-FM2500164 Revision NIC.Instructions for Continued Airworthiness GC92380000-IFCA.</td>
<td>Restricted to ASN 9238 and Refer to the Limitations sections of the required AFM supplement and to the referenced FCOMS.WLAN System is compatible only to IEEE 802.11(b).</td>
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<td>TCCA STC (SA05-40) FOR CABIN</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Customised Interior Completion - ACSN 9241.</td>
<td>TCCA SOC letter C-08-365 dated 30 April 2008. Aero Consulting Services Configuration Definition List EQ25000 Revision AK dated 29 November 2007. Bombardier Airplane Flight Manual Supplement G-FM2500405 Revision NIC or later approved revision.Instruction for Continued Airworthiness GC92410000-IFCA.</td>
<td>Restricted to ASN 9241 and Refer to the Limitations sections of the required AFM Supplement.</td>
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<td>TCCA LIMITED STC C-LSA07-278/D</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Customised Interior Completion - ACSN 9243.</td>
<td>TCCA STC C-LSA07-276/D. Issue 1, dated 06-12-2007. TCCA SOC letter C-07-0704 dated 7-12-2007. Aero Consulting Services Modification Data Summary F4570000 Revision NIC dated 06-12-2007. Bombardier Airplane Flight Manual Supplement G-FM2500162 Revision NIC or later approved revision.Instruction for Continued Airworthiness GC92430000-IFCA.</td>
<td>Refer to the Limitations section of the required AFM Supplement and to the referenced FCOMS.WLAN system is compatible only to IEEE 802.11 (B).</td>
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<td>JAMCO AMERICA, INC.</td>
<td>FAA STC ST01460SE</td>
<td>BOEING 767-300</td>
<td>Oxygen Module Installation.</td>
<td>JAMCO Master Drawing List IA-122ML, Revision C, 31/08/2004.</td>
<td>Reconfiguration of Passenger seats as per FAA STC SA7032NM-D must be installed concurrently. This FAA STC must be approved by EASA directly or by Grandfather Rights before installation on a EASA member state aircraft.</td>
<td>10/01/2008</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST10312SC-D</td>
<td>G150</td>
<td>Installation of a VHF COM Antenna.</td>
<td>IFCA: GA34300A000.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will not introduce any adverse affect upon the airworthiness of the product. The installation of this modification by third persons is prohibited.</td>
<td>22/11/2007</td>
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<td>FAA STC ST10315SC-D</td>
<td>GULFSTREAM G150</td>
<td>Installation of a 3rd VHF COM with Data link.</td>
<td>IFCA: GA31300A000AFMS: GA31300M000</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is prohibited.</td>
<td>22/11/2007</td>
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**List of EASA Supplemental Type Certificates**

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<td>BOMBERDIER COMPLETION CENTRE INC</td>
<td>TCAS STC SA-03/113</td>
<td>BOMBERDIER CL-600-2B16 (604)</td>
<td>Installation, complete Custom Interior</td>
<td>Transports Canada STC SA-03/113 issue 1 dated December 19 2005 Aero Consulting Services Ltd. Configuration Definition List D817000 Revision A.Aero Consulting Services Ltd. APM Supplement D817000. Revision A or later approved revision.</td>
<td>Restricted to ASN 5623.</td>
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<td>GKN AEROSPACE CHEM-TRONICS Inc</td>
<td>FAA STC SA8581NM</td>
<td>MD-88</td>
<td>FAA STC SA8581NM *Installation of Astech/MD-80 Series tailpipe, Rev NIC dated July 10, 1992 or later FAA approved revision.</td>
<td>This approval should not be extended to other specified airplanes of this model series on which other previously approved modifications are incorporated, unless it is determined that the interrelationship between this change and any other previously approved modifications will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/03/2008</td>
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<td>CONCORDE BATTERY CORPORATION</td>
<td>FAA STC NO. ST01173W</td>
<td>LEARJET 45</td>
<td>Installation of Concorde RG-443 Lead Acid Battery.</td>
<td>None.</td>
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<td>S-TEC CORPORATION</td>
<td>FAA STC SA9001AC-D</td>
<td>CESSNA T310R</td>
<td>Installation of S-TEC System SA-650 Two Axis Automatic Flight Guidance System. Model ST-694, also when modified per FAA/STC as listed in 3.5 (Limitations and Conditions).</td>
<td>FAA approved Bulletin No. 694, Revision 4, dated 11-May-2001, or later approved revision FAA approved Master Drawing. List No. 9079, Revision D, dated 11-May-2001, or later approved revision (28 Volt System)/FAA/DAS Approved Supplemental Flight Manual, PN 991179, Revision 1, dated 11-May-2001, or later.</td>
<td>This STC is also eligible on variants 310K, 310L, and 310N when modified per STC SA412SO (300 H.P. Calembur Conversion). This STC is also eligible on variants T310P, and T310G when modified per STC SA750SW (RAM Modification). Installer must check availability of EASA-approval for one or more mand...</td>
<td>22/11/2007</td>
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<td>EASA/IM.A.S.02629</td>
<td>COLUMBIA AVIONICS INC.</td>
<td>FAA STC ST01162WI</td>
<td>CESSNA 515</td>
<td>Installation of S-TEC System SA-650 Two Axis Automatic Flight Guidance System. Model ST-694, also when modified per FAA/STC as listed in 3.5 (Limitations and Conditions).</td>
<td>FAA approved Bulletin No. 694, Revision 4, dated 11-May-2001, or later approved revision FAA approved Master Drawing. List No. 9079, Revision D, dated 11-May-2001, or later approved revision (28 Volt System)/FAA/DAS Approved Supplemental Flight Manual, PN 991179, Revision 1, dated 11-May-2001, or later.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previous or previously approved modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by the...</td>
<td>19/03/2009</td>
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<td>EASA/IM.A.S.02630</td>
<td>COLUMBIA AVIONICS INC.</td>
<td>FAA STC ST01084WI</td>
<td>HS125 SERIES 700A/B, HS125 SE. -403B, F400B, - F400B, HS125 SE. 400B, -401B, -401B, HS125 SE. 400B, -400B/2, -400B/3, - F400B, HS125 SE. 400A, -600B, -600B, -600B, -600B, HS125 SE. 400A, -600B, -600B/3, - F400B, HS125 SE. 400A, -600B, -600B, -600B, -600B, HS125 SE. 400A, -400A, HS125 SE. 400A, -400A</td>
<td>Installation in accordance with Columbia Avionics, Inc. Master Document List No. 30/140565, Revision 2, dated May 31, 2002, or later approved revisions. Operation in accordance with FAA approved Supplement 30/140565, dated June 7, 2002 or later Approved Revisions.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/06/2008</td>
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<td>EASA.IM.A.S.02631</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03530AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of AirCell Axxess II Iridium Satellite Telephone.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List GC120036004 Rev C, dated 1 April 2008, or later approved revisions.</td>
<td>Aircraft Serial 5180 and subsequent.</td>
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<td>EASA.IM.A.S.02632</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03532AT-D</td>
<td>GULFSTREAM GV-X</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4104.</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Master Index List, GCA14914104 Rev C, dated February 27, 2008, or later approved revisions with accordance EASA ED Decision 2004/494/EC (or subsequent revisions of this decision).</td>
<td>1. Aircraft Serial number 4104 only. 2. The following STCs are a required part of this STC: ST03027AT-D2...ST02987AT-D2...ST03079AT-D2...ST03098AT-D2...ST03238AT-D2...ST03328AT-D2...ST03338AT-D3... It is the installer's responsibility to ensure these STCs are all EASA approved.4.</td>
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<td>EASA.IM.A.S.02633</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03532AT-D</td>
<td>GULFSTREAM GV-X</td>
<td>Installation of AirCell Axxess II Iridium Satellite Telephone.</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Index List, GC120036004 Rev C, dated April 14, 2008, or later approved revisions.</td>
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<td>EASA.IM.A.S.02634</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST03532AT-D</td>
<td>GULFSTREAM GV-S</td>
<td>Aircraft Serial No 5176 completion and cabin interior installation</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Index List No. GC52 2046083, Revision A, dated March 24, 2008, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 2044001, Original release, dated March 24, 2008, or later approved.</td>
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<td>EASA.IM.A.S.02635</td>
<td>THE BOEING COMPANY - WICHITA DIVISION</td>
<td>FAA STC SA/123CE-D AMENDED 19</td>
<td>BOEING 747-28B</td>
<td>Conversion of Passenger Airplane to dedicated Special Freighter with Main Deck Side Cargo Door in accordance with Boeing Drawing List document D3-8857, Section V, Appendix 31, Rev C</td>
<td>This approval is limited to airplane model B747-28B8 S/N 22482 and S/N 22485. Boeing Commercial Airplane Groups - Wichita Division. Airplane Flight Manual Supplement dated 9 September 1991 or revision for AFM D6-13703.385 or later approved revisions of this document approved by EASA in accordance with FAA approved revision.</td>
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<td>EASA.IM.A.S.02636</td>
<td>ROYAL ATLANTIC AVIATION, INC.</td>
<td>FAA STC SA/2016AT</td>
<td>CESSNA 441 SERIES</td>
<td>Ground Recognition Strobe Light System</td>
<td>Installation law. Master Drawing List MDL RAA- MDL-1, Rev. A, or later FAA approved Revisions and MDL RAA-INST-1, Rev. B, or later approved Revisions. To be maintained law. Document No RAA-ICA-1, Rev. IR, or later approved Revisions. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification b...</td>
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<td>EASA.IM.A.S.02637</td>
<td>B&amp;B PARTNERS</td>
<td>FAA ST03483AT</td>
<td>SEE EASA-APPROVED MODEL LIST</td>
<td>Installation of a portable beverage dispensing system, PIN B43-FA 001</td>
<td>- FAA STC ST03483AT dated 24 September 2007. B&amp;B Partners Assembly Drawing B43-FA, dated 21 December 2006. - Master Document List No. 6009, dated 14 August 2007. - Instructions for Continued Airworthiness for PIN B43-FA-001 Rev. Added 9 August 2007. - Drawing No. 1178 on *Galley Interface...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. This modification b...</td>
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| EASA.IMA.S.02639 | DELTA ENGINEERING CORPORATION     |            | FAA STC ST02630NY  | BOEING 777-200 | Heath Tecnica Interior                                                    | FAA STC ST02630NY Heath Tecnica Engineering Master Data List 0143-0204-2159
Revision A dated November 25, 2008 or later approved revisions | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons... | 15/12/2008 | Active  | 1016332 |
| EASA.IMA.S.02639 | BOMBARDIER COMPLETION CENTRE INC  |            | TCCA G-LSA01-007 ISSUE 6 | BOMBARDIER BD-700-1A10 | Interior and Avionics Installation.                                      | TCCA ScV 5010-0-07-0746 dated 13 May 2008Decca Aviation Engineering Ltd MDL No. MDL0004, Revision E, dated 10 April 2002. The following DECA Aviation Engineering Ltd Flight Manual Supplemental are required:A. interior and Avionics Interior and Avionics Installation. | 1. Limited to AS N90042. Compliance with Bombardier Aerospace Airworthiness Limitations GC-9045000-AWL, dated July 15, 2002, is mandatory. 3. Occupancy: Maximum number of passengers Twelve (12); Maximum number of occupants including flight crew Sixteen (16). The Public Address... | 22/05/2008 | Active  | 1016333 |
| EASA.IMA.S.02640 | S.I.P. TECHNICAL SERVICES            |            | FAA STC NO. ST96465C | BOEING DC-9-83 (MD-83) BOEING DC-9-92 (MD-82) BOEING DC-9-91 (MD-81) | Installation of required sensor and Interface for existing DFDR and FDAU to meet FAR 121.341. | Master Data List 3275-001-561 Rev. D dated December 5, 2002 | Master Data List 3275-001-561 Rev. D dated December 5, 2002 | 10/06/2006 | Active  | 1016331 |
| EASA.IMA.S.02642 | BOMBARDIER INC.                     |            | TCCA STC NO. SA04-112 ISSUE 3 | BOMBARDIER BD-100-1A10 (CHALLE) | Installations limited to Gulfstream G200, S/N 163 only.                  | This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20141. | This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20141. | 18/12/2007 | Active  | 1016333 |
| EASA.IMA.S.02643 | GULFSTREAM AEROSPACE CORPORATION     |            | FAA STC ST09623SC-D  | GULFSTREAM 200 | Installation of Dual UNS 1C + upgraded FMS i.a.w. Index List CEB11188001 | 1. AFMS CEB11184M001, IFCA CEB11184A001 | 1. For A/C S/N004 through 051 only. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The install... | 18/12/2007 | Active  | 1016334 |
| EASA.IMA.S.02644 | THOMMEN AIRCRAFT EQUIPMENT AG       |            | FAA STC ST106855C  | LEARJET 60 | Installation of Revue Thommen 3A43 Standby Altimeter. | As noted in MDL346003, rev A. | As noted in MDL346003, rev A. | 12/03/2008 | Active  | 1016335 |
| EASA.IMA.S.02645 | GULFSTREAM AEROSPACE CORPORATION     |            | FAA STC ST10329SC-D  | GULFSTREAM 200 | Installation of a forward RH Galley with a convertible high Temp Oven/Microwave Oven Configuration. | Installation of FWD right hand Galley with a convertible high Temp Oven/ Microwave oven configuration in accordance with Index list GA223288002, Instruction for continued Airworthiness GA22328A001, dated 19.07.2007 | Installation of a forward RH Galley with a convertible high Temp Oven/Microwave Oven Configuration. | 17/04/2008 | Active  | 1016336 |
| EASA.IMA.S.02646 | THOMMEN AIRCRAFT EQUIPMENT AG       |            | FAA STC ST106845C  | LEARJET 45 | Installation of Revue Thommen 3A43 Standby Altimeter. | As noted in MDL3445002, rev A (or later approved revisions). | As noted in MDL3445002, rev A (or later approved revisions). | 12/03/2008 | Active  | 1016337 |

**List of EASA Supplemental Type Certificates**

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## List of EASA Supplemental Type Certificates

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<td>EASA.IM.A.S.02647</td>
<td>SIMMONDS PRECISION PRODUCTS, Inc.</td>
<td>FAA STC ST000280</td>
<td>AIRBUS A300</td>
<td>In-Tank Fuel Quantity Indicating System (FAA STC ST000280 validation). This STC installs, within the aircraft fuel tanks, a new FQIS sensors and the associated wire harnesses.</td>
<td>Master Drawing and Data List Document FMS-012 – Airplane Flight Manual Supplement for Citation 500 and Citation 550.</td>
<td>Revision D, dated 02 March 2007 or any later EASA approved revision or instruction for Continued Airworthiness.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/10/2008</td>
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<td>EASA.IM.A.S.02650</td>
<td>SIERRA INDUSTRIES Ltd.</td>
<td>FAA STC SA174RM</td>
<td>CESSNA 550</td>
<td>Installation of an extended width cabin door.</td>
<td>Master Doc. List MDL 91500, Report 151-1 “Cessna Citation Extended Door Structural Analysis”, Rev. A dated 2-25-88.</td>
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<td>EASA.IM.A.S.02654</td>
<td>AERODYME CORPORATION</td>
<td>FAA STC SA0246B0</td>
<td>114, 114A</td>
<td>Installation of Longstanding IO-540-T4B5 engines.</td>
<td>FAA approved Aircraft IO-540 engine Supersede Installation Instructions, Doc. No. AD725-01, dated 02 November 2006 or later approved version.</td>
<td></td>
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<td>27/11/2007</td>
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<td>EASA.IM.A.S.02660</td>
<td>FLYHT AEROSPACE SOLUTIONS Ltd.</td>
<td>REV. 2</td>
<td>TCCA STC SA07-8</td>
<td>BOEING 737-700 SERIES</td>
<td>AFIRS 220 Automated Flight Information Reporting System installation with optional Global Voice and Global Text Satellite Communication.</td>
<td>AFMS, Wiring Diagram Manual or later revisions of this document approved by EASA in accordance with EASA ED Decision 20041002CF (or subsequent revisions of this decision).</td>
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<td>22/05/2008</td>
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<td>EASA.IM.A.S.02662</td>
<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>FAA STC ST2685AT</td>
<td>SAAB SF340A</td>
<td>Installation of FA2190CVF Cockpit Voice Recorder in Saab 340 series.</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST10313C-0</td>
<td>G150</td>
<td>Installation of an XM Antenna.</td>
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<td>ICA GA34607A0</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
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<td>EASA.IM.A.S.02671</td>
<td>B&amp;G SPECIALITY PRODUCTS</td>
<td>FAA SA00846WI</td>
<td>T210L, T210M, T210N 210L, 210M, 210N, P210N</td>
<td>Addition of Vacuum Pump drive alternator and regulator with associated controls.</td>
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<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.A.S.02672</td>
<td>HAWKER BEECHCRAFT SERVICES</td>
<td>FAA STC SA01414WI</td>
<td>HAWKER BEECHCRAFT G58 (BARON)</td>
<td>Modification to install a Honeywell KR 87 Automatic Direction Finder (ADF) System and the Honeywell KA-446 Antenna interfaced with the previously installed Garmin 1000 Avionics System.</td>
<td></td>
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<td>Hawker Beechcraft Services • Master Drawing List No. RA500630364 Revision 1.0 dated 11 Oct. 2006 or later FAA approved revision • Airplane Flight Manual Supplement RA500630379 dated 6 Oct. 2006 or later FAA approved revision.</td>
<td>None</td>
<td>27/05/2009</td>
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<td>EASA.IM.A.S.02673</td>
<td>HAWKER BEECHCRAFT SERVICES</td>
<td>FAA STC SA01455WI</td>
<td>HAWKER BEECHCRAFT G36 (BONANZA)</td>
<td>Modification to install a Honeywell KR 87 Automatic Direction Finder (ADF) System and the Honeywell KA-446 Antenna interfaced with the previously installed Garmin 1000 Avionics System.</td>
<td></td>
<td></td>
<td>Hawker Beechcraft Services • Master Drawing List No. 002-3450- L01-00 Revision 1.0 dated 26 Sept. 2007 or later FAA approved revision • Airplane Flight Manual Supplement 002-3450-A01-00 dated 15 Oct. 2007 or later FAA approved revision.</td>
<td>None</td>
<td>27/05/2009</td>
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<td>EASA.IM.A.S.02675</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SA00941SE</td>
<td>Cessna T210N Cessna T210M Cessna T210L Cessna T210K Cessna T210G Cessna T210N Cessna 2100G, H, J, K, L, M, N,</td>
<td>Installation of vortex generators on the wing and tail surfaces.</td>
<td></td>
<td></td>
<td>Installation must be in accordance with Drawing Package MA2159 Rev. IR dated 18 April 2001 and Installation Manual MA2157 Rev. IR dated 03- July-2001, or later Approved Revisions. Maintenance must be in accordance with Micro Aerodynamics, Inc. Instruction for Continued Airworthiness related to the FAA S...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.A.S.02676</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA O-LSA07-261D ISSUE 1 07</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Cabine Floor Structure Modification &amp; Decompression Panel Relocation.</td>
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<td>TCCA Statement of compliance reference 5018-O-07-0790 dated 04 April 2008. DECA Aviation Modification Summary MS21759 Revision 1 dated 18 September 2007 or later approved revision Instruction for Continued Airworthiness ICA01759 Revision 1 dated 22 October 2007 or later approved revision.</td>
<td>05/05/2008</td>
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<td>EASA.IM.A.S.02678</td>
<td>LEARJET Inc.</td>
<td>FAA 8191NM-D</td>
<td>BOMBARDIER CL-600-2B16 (604)</td>
<td>Installation of a Safe Flight Enhanced Autopower System.</td>
<td>Learjet Inc. MDL No. DC223050/1 at revision U, dated 26 April 2007<em>Learjet Inc. AFMS No. BA60040125 at revision NC, dated 28 April 2007</em> required for -3 configuration; or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions).</td>
<td>The -3 configuration requires the installation of FAA STC 8191NM-D-3 configuration.</td>
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<td>EASA.IM.A.S.02684</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA C-LSA08-014/D</td>
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<td>BD-100-1A10</td>
<td>AIRCRAFT COMPLETION, 10 PASSENGERS INTERIOR CONFIGURATION/EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA08-014/DIssue 1, dated 20th February 2008.</td>
<td>Transport Canada Limited Supplemental Type Certificate C-LSA08-014/DIssue 1, dated 20th February 2008.</td>
<td>This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20179.</td>
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<td>EASA.IM.A.S.02685</td>
<td>ARMSTRONG AEROSPACE Inc dba ASTRONICS</td>
<td>FAA ST02501CH</td>
<td>AIRBUS B4-605R A360 B4-603</td>
<td>Activation of Honeywell Runway Awareness and Advisory System (RAAS) (Validation of the FAA STC n° ST02501CH)The STC consists in activating the RAAS function on EGPRS Master Data List, ref: OC104-5029-00 Revision B, dated 08 November 2007. (or any later approved revision of this document) Certification Review Item F-01, issue IR, dated 18 February 2008; Aircraft Flight Manual Supplement, ref: DC303-5029-00, dated 16 November 2007; Instruction for Continued Airworthiness for Installation of A 115 cubic foot Oxygen Cylinder.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is not permitted.</td>
<td>13/05/2008</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST103195C-D</td>
<td>G150</td>
<td>Installation of LaserRef V IRS.</td>
<td>ICA GA31313A000AFMS GA31313M000</td>
<td>Instruction for Continued Airworthiness for Installation of a 115 cubic foot Long Range Oxygen Cylinder (Validation of the FAA STC n° ST032209C-D) The STC consists in installing a 115 cubic foot oxygen cylinder in accordance with Index List GA32708B000, Rev B, dated 13.09.2006 or later versions of the above listed documents approved by EASA in accordance with EASA ED Decision 2007/04/CF (or subsequent revisions of this decision).</td>
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<td>26/03/2008</td>
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<td>EASA.IM.A.S.02687</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
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<td>G150</td>
<td>Installation of a 115 cubic foot oxygen cylinder in accordance with Index List GA32708B000, Rev B, dated 13.09.2006 or later versions of the above listed documents approved by EASA in accordance with EASA ED Decision 2007/04/CF (or subsequent revisions of this decision).</td>
<td>FAA STC ST02122LA, issued 4th March 2008</td>
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<td>AIRCRAFT TECHNICAL DEVELOPMENT Inc dba AEROCOM ENGINEERING COMPANY</td>
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<td>Installation of Aerocom Engineering Company (AEC) On-Board AirSAR</td>
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<td>EASA.IM.A.S.02693</td>
<td>HAWKER BEECHCRAFT CORPORATION (HBC)</td>
<td>FAA STC SAD1986E</td>
<td>BEECH 300C BEECH 300 BEECH 200CT BEECH 200C BEECH 200</td>
<td>Reduced Vertical Separation Minimum Qualification Document Group Aircraft Certification Honeywell Beechcraft Models B200/B200C/B200CT/B200C/GT/B300C (Super King Air)</td>
<td>Aero-Mech AM/AFM/AIR-GULFSTREAM-4979 Airplane Flight Manual Supplement for Hawker Beechcraft Models B200/B200C/B200C/GT/B200C/GT</td>
<td>Aero-Mech AM/AFM/AIR-GULFSTREAM-4979 Airplane Flight Manual Supplement for Hawker Beechcraft Models B200/B200C/B200C/GT/B200C/GT</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons may release the product. The installation of this modification by third persons will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>FAA STC ST03524AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial Number 5178.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915176, Rev. D, dated 1 May 2008, or later approved revisions.</td>
<td>Aircraft Completion for Aircraft Serial Number 5178.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons may result in no adverse effect upon the airworthiness of the product. The installation of this modification by third persons will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03525AT-D</td>
<td>GULFSTREAM GV-X</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial Number 4103.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915176, Rev. D, dated March 31, 2008 or later approved revisions.</td>
<td>Aircraft Completion for Aircraft Serial Number 4103.</td>
<td>Aircraft Completion for Aircraft Serial Number 4103.</td>
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<td>FAA STC ST03526AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial Number 5174.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915176, Rev. D, dated 19 March 2008 or later approved revisions.</td>
<td>Aircraft Completion for Aircraft Serial Number 5174.</td>
<td>Aircraft Completion for Aircraft Serial Number 5174.</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03527AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Airplane Completion STC on Aircraft Serial Number 5167.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915176, Rev. C, dated 1 January 2008.</td>
<td>Aircraft Completion STC on Aircraft Serial Number 5167.</td>
<td>Aircraft Completion STC on Aircraft Serial Number 5167.</td>
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<td>FAA STC ST03528AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial Number 5174.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915176, Rev. E, dated 13 May 2008 or later approved revisions.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915176, Rev. E, dated 13 May 2008 or later approved revisions.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915176, Rev. E, dated 13 May 2008 or later approved revisions.</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03529AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Airplane Completion STC on Aircraft Serial Number 5166.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915176, Rev. C, dated 19 March 2008 or later approved revisions.</td>
<td>Aircraft Completion STC on Aircraft Serial Number 5166.</td>
<td>Aircraft Completion STC on Aircraft Serial Number 5166.</td>
<td>6/11/2008</td>
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**List of EASA Supplemental Type Certificates**

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<th>EASA STC No.</th>
<th>STC Holder</th>
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<td>EASA.IM.A.S.02701</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC SA04-113</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Customized interior Completion - ACSN 9254.</td>
<td>TCCA SOC Letter C-06-0342 dated 06 May 2008. Aeronautical Consulting Services Configuration Deviation List D645000 Revision BK dated 11 April 2008. Bombardier Airplane Flight Manual Supplement G-FM25004001 Revision N/C. Instruction for Continued Airworthiness GC92540000-IFCA.</td>
<td>Restricted to ASN 9254 and 1. Prior installation and approval of a Collins Control Display Unit, type CDU-4100, is a prerequisite to this STC. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will intro...</td>
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<td>EASA.IM.A.S.02704</td>
<td>BASIC AIRCRAFT PRODUCTS, INC.</td>
<td>FAA ST00702AT</td>
<td>210</td>
<td>Installation of Basic Type mBAE-14/28 Model 2025 Deployable/Driven Alternator System</td>
<td>Basic Aircraft Products, Inc. Installation Kit No. 45-2525-OPL, Rev. A, dated 26-08-1994 and Installation Manual No. 80-2025-101, dated 24-08-1994 or later approved revisions.</td>
<td>This approval should not be extended to other aircraft of this model on which other previously approved modifications will produce no adverse effect upon the airworthiness of that aircraft. EASA approved Supplemental Airplane Flight Manual or EASA approved Airplane Flight Manual Supplement, dated 28-03-.</td>
<td>18/01/2008</td>
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<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCAC STC SA01-156</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Multi-Function Ladder Installation.</td>
<td>Transport Canada Statement of Compliance ref. NAV 5012-01-0610 Transport Canada STC No. SA01-156 Issue 1 dated 10 December 2001Eesan Technologies Inc Master Data List No. MDL-020X-117 Revision A dated 20 December 2001 or later approved revision</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>IMA.02707</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>STO2084LA-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of a Smiths Industries Flight Computing Device Recorder.</td>
<td>Install in accordance with Gulfstream Aerospace Report GCS2008M071 Revision Original, dated April 18, 2008, or later approved revision</td>
<td>Aircraft Serial Number 5175 only.</td>
<td>13/05/2008</td>
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<td>IMA.02708</td>
<td>GOODRICH LIGHTING SYSTEMS GmbH</td>
<td>FAA STC STO2050LA</td>
<td>2. DC-10-15/40-40F, MD-11 1. MD-11-11F 1. DC-10-10/30-30F</td>
<td>Anticollision Recognition Lighting System (ARLS) in accordance with Master Drawing List, Issue No. R-386-00 dated 5 December 2007 or later EASA-approved revision.</td>
<td>Data package containing the following documents, and additional supporting data: 1. FAA STO2050LA A2 FAA Letter dated 27th April 2007. Project Specific Certification Plan R-386 Project no. ST11125LA- Td. 12 May 2004. Master Drawing List R-386-00 d.d. 9 December 2007. E/0 Installatio...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>FLYHT AEROSPACE SOLUTIONS Ltd.</td>
<td>TCCA STC SA07-84</td>
<td>BOMBARDIER CL-600-2B19 (SERIES)</td>
<td>AFIRS 220 Automated Flight Information Reporting System installation with Optional Global Voice and Global Text Satellite Communication</td>
<td>TCCA Statement of Compliance ref. C-07-0944 dated 02 July 2006AMS Master Document List, Ref. 121-0032 Rev. - dated 29 June 2007 or later approved revisionAMS Installation Instructions, Ref. 151-0027 Rev. - dated 27 March 2007 or later approved revision</td>
<td>AMS Instructions for Continued Airworthiness, Ref. 153-0206Rev. A or later approved revision. Compliance with the approved Airworthiness Limitations contained therein is mandatory.</td>
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<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC SA01527CH</td>
<td>PILATUS PC-12/47 PILATUS PC-12/45 PILATUS PC-12</td>
<td>Installation of a Honeywell KMH 880 Multi-Hazard Awareness System or a Honeywell KTA 870 Traffic Advisory System.</td>
<td>FAA approved Electronic Cable Specialists Master Data List PC-12, PC-12/45, PC-12/47, Installation of IHAS, ECS-200836 Revision L dated August 29, 2006, or later FAA approved revision. FAA approved Airplane Flight Manual Supplement as listed in Appendix A of Electronic Cable Specialist Master Data...</td>
<td>Limitations as per Master Data List PC-12, PC-12/45, PC-12/47, Installation of IHAS, ECS-200836 Revision L, dated August 29, 2006, or later FAA approved revision, Part III. Limitations</td>
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<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC STO1479CH</td>
<td>BOEING 737-300-900 SERIES</td>
<td>Installation of a Smiths Industries Flight Management System Data Bus Portable Computing Device Recorder.</td>
<td>Master Data List, Doc. No. ECS-200447 Rev. K dated June 26, 2007FAA Approved Flight Manual Supplement Doc. No. ECS-200875, Revision B, approved October 5, 2007 is required on board for MDL configuration 1, 2, and 3Airplane Flight Manual Supplement Doc. No. ECS-205138, Revision N/ C, approved October...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>SIERRACIN/SYLMAR CORPORATION</td>
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<td>FAA STC</td>
<td>BOMBARDERDHC-8-300 SERIES BOMBARDERDHC-8-200 SERIES BOMBARDERDHC-8-100 SERIES</td>
<td>Installation of Sierracin/Sylmar Windshield Assembly P/N 802620-111-12, Ref.: FAA STC ST0164NY.</td>
<td>Modification Data Summary D641000, Revision N/C, dated 4 November 2003*Instructions for Continued Airworthiness/ Maintenance to be performed with Component Maintenance Manual 56-10-12, Rev. A, dated 3 September 2003** or later rev.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>SIERRACIN/SYLMAR CORPORATION</td>
<td>REV.1</td>
<td>FAA STC</td>
<td>BOMBARDERDHC-8-400 SERIES BOMBARDERDHC-8-300 SERIES BOMBARDERDHC-8-200 SERIES BOMBARDERDHC-8-100 SERIES</td>
<td>Installation of Sierracin/Sylmar Windshield Assembly P/N 802620-096-10, Ref. FAA STC ST0169NY</td>
<td>Modification Data Summary D642000, Revision N/C, dated 27 October 2003*Instructions for Continued Airworthiness/ Maintenance to be performed with Component Maintenance Manual 56-10-12, Rev. A, dated 3 September 2003** or later rev.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
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<td>FAA STC</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4106.</td>
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<td>Applicable to Aircraft Serial Number 4106 only. The following STCs are required as part of this STC: FAA STC ST03238AT-D, ST02987AT-D, ST03027AT-D, ST03047AT-D, ST03239AT-D, ST02989AT-D, ST02979AT-D, ST02989AT-D, ST03029AT-D, ST03029AT-D.</td>
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<td>DUNCAN AVIATION, INC.</td>
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<td>FAA STC</td>
<td>LOCKHEED 1011-385-1</td>
<td>Installation of Dual IS&amp;S Air Data Attenuator System along with two inch standby pneumatic altimeter (FAA STC ST0031464-D).</td>
<td>Master Document List 100553800, Rev. D<em>Supplement to Aircraft Flight Manual (AFM) Doc. 40028020, Rev. G</em>Supplement to AFM/IN 5988 only Doc. 050822006, Rev. 0*Instructions for Continuous Airworthiness Doc. 44503500, Rev. A</td>
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<td>ROCKWELL COLLINS, Inc.</td>
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<td>FAA STC</td>
<td>G200, G105, HAWKER 800XP BD-100-1A10, CL-600-2B16</td>
<td>Modification of Collins FMS-5000 or FMS 6000 and Installation in Part 25 Airplanes i.e. Drawing index as listed in AML Nr. ST01452WI-D Rev. pr master approved revisions approved by EASA i.e. EASA ED decision 2004/04/CF (or subsequent revision of this decision).</td>
<td>Approved model List No. 701452694-Disinstallation instructions RCA-2447.</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>EASA.IM.A.S.02725</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STD ST00649W-D</td>
<td>BOEING 747-300 BOEING 747-200F BOEING 747-200C BOEING 747-200B</td>
<td>Modification of Collins 618M-3A VHF COMM R/T System Installation to Collins VHF-800B VHF COMM R/T System. Data package containing the following documents, and additional supporting data: 1. Collins Drawing Index 902-4304-001 2. FAA STC ST00649W-D 3. Rockwell-Collins document R98847: STC Certification Plan for Collins VHF-800B 4. 33 kHz VHF Communications Receiver/Transmitter for Boeing 747-200B and 300...</td>
<td>This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>20/02/2008</td>
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<td>EASA.IM.A.S.02727</td>
<td>ROCKWELL COLLINS, Inc.</td>
<td>FAA STC SA01453WI-D</td>
<td>PIAZZO P 180 CESSNA 525A CESSNA 525 BEECH B330C BEECH B300 BEECH B200 BEECH 390 BEECH 300</td>
<td>Modification of a Collins Flight Management System type FMS-3000 Installed in Part 23 Aircraft. Installation in accordance with Collins Drawing Index as listed in AML No EASA.IM.A.S.02727 at latest approved revision. Operation in accordance with Collins Flight Manual Supplement as listed in AML No EASA.IM.A.S.02727 at latest approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>LEARJET Inc.</td>
<td>FAA ST001680</td>
<td>BOMBARDIER BD-700-1A11 BOMBARDIER BD-700-1A10</td>
<td>Installation of a main entry door rain deflector channel.</td>
<td>FAA STC ST001680 dated 21 November 2007 FAA SOC dated 06 May 2008 Bombardier Aerospace MDL No. BAS9701703, Revision A, dated 11 November 2007 or later approved revision.</td>
<td>08/06/2008</td>
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<td>1001640</td>
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<td>EASA.IM.A.S.02731</td>
<td>HONEYWELL INTERNATIONAL INC.</td>
<td>FAA STC ST00381SE</td>
<td>BOEING 757</td>
<td>Installation of Enhanced Ground Proximity Warning System inB-757 Series Aircraft Approval applies to aircraft equipped with a flight management system, which meets AC30-45A, or an EGPWS with a GPS based navigation system or an EGPWS equipped with an internal GPS and a GPS antenna as defined in MAA...</td>
<td>Flight Manual Supplements approved by FAA096-4199-012, 060-4189-015, 060-4189-029, 060-4189-02050-4189-05 5, 060-4189-064, 060-4189-065, 060-4189-06500-4189-08 1, 060-4189-082, 060-4189-079, 060-4189-117Operating &amp; Service Instructions*: 060-4199-180*See Applicable documents in Honeywell STC...</td>
<td>11/04/2008</td>
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<td>Description</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.02734</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC</td>
<td>SA00928AC-D</td>
<td>PIPER PA-31-325 PIPER PA-31</td>
<td>Installation of S-TEC SYSTEM 55/55X Two Axis Automatic Flight Guidance System, Model ST.751, also when modified per FAA STC as listed in 3.5 (Limitations and Conditions).</td>
<td>FAA approved Bulletin No. 851, Revision 3, dated 3-12-01 or later FAA approved revisions (28 Volt System)/FAA/DAS Approved Pilot's Operating Handbook and/or Airplane Flight Manual Supple...</td>
<td>This STC is also eligible on Models PA-31 when modified per STC SA09750 (350 hp Colmillo Conversion)/This STC is also eligible on Model PA-31-325 when modified per STC SA00485E (Vortex Generators System)/Prior to installation of this modification the installer must determine that the interrelati...</td>
<td>22/05/2008</td>
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<td>EASA.IM.A.S.02735</td>
<td>GULFSTREAM AEROSPACE CORPORATION.</td>
<td>FAA STC</td>
<td>ST100346C-D</td>
<td>G150</td>
<td>Installation of a Lavatory Smoke Detection system.</td>
<td>ICA GAI331455A000AFMS GA31455M000</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>18/03/2008</td>
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<td>EASA.IM.A.S.02737</td>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD.</td>
<td>TCCA STC Q-LSA08-001D</td>
<td>BOEING</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Replacement of conference table in aircraft S/N 9197, aft cabin.</td>
<td>Transport Canada LSTC No. LSA08-001D Issue 1 dated 15 January 2008/Transport Canada Statement of Compliance ref: NAKOS12-08-0003 dated 18 January 2008/MDL2714202, Master Document/List - Conference Table Replacement on Aircraft S/N 9197, revision NC, dated January 15, 2008/TL2714204, Time Limits a...</td>
<td>Restricted to ASN 9197/With this modification incorporated Aft LSH forward facing double seats must not be occupied during taxi/take off and landing.</td>
<td>19/03/2008</td>
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<td>EASA.IM.A.S.02738</td>
<td>ANDRA INTERNATIONAL LLC</td>
<td>FAA STC</td>
<td>ST01388LA</td>
<td>747-200, 747-300, 747-400 3,777-200, 777-300 2,767-200, 767-300, 767-400ER</td>
<td>Installation of Power Drive Unit (Cargo System).</td>
<td>747 Series Ancra Controlled Data Lists, DL92001, Revision E, 02/07/2007 DL92002, Revision E, 02/07/2007 DL92003, Revision B, 02/07/2007 Series Ancra Controlled Data List, DL92001, Revision E, 02/07/2007 DL2002, Revision E, 02/07/2007</td>
<td>Restricted to ASN 9197/With this modification incorporated All LSH forward facing double seats must not be occupied during taxi/take off and landing.</td>
<td>31/03/2008</td>
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<td>EASA.IM.A.S.02743</td>
<td>JET AVIATION ST. LOUIS, Inc.</td>
<td>FAA STC</td>
<td>ST00794W</td>
<td>BOMBARDIER CL-600-2B16 BOMBARDIER CL-600-2A12 (CL-601) BOMBARDIER CL-600-1A1 (CL-600)</td>
<td>Installation of Emergency Lighting System.</td>
<td>Midcoast Aviation Certification Drawing List 34-415-G001 Revision E dated March 20, 1999 or later approved revisions.</td>
<td>This STC must be installed with Cabin interior arrangement per EASA STC No. EASA.IM.A.S. 02745 and an oxygen system per EASA STC EASA.IM.A.S. 02744 or other approved interior arrangement and oxygen system.</td>
<td>20/03/2008</td>
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<td>Limitations/Conditions</td>
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<td>EASA.IM.A.S.02744</td>
<td>JET AVIATION St. LOUIS, Inc.</td>
<td>FAA STC</td>
<td>ST00758W</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Installation of Passenger Oxygen System.</td>
<td>Microblog Aviation Certification Drawing List 34-261G001 Revision B dated March 15, 1999, or later approved revision.</td>
<td>This STC must be installed with Cabin interior arrangement per EASA STC No. EASA.IM.A.S. 02745 and an emergency lighting system per EASA STC No. IM.A.S.02743 or other approved interior arrangement and emergency lighting system.</td>
<td>20/02/2008</td>
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<td>EASA.IM.A.S.02745</td>
<td>JET AVIATION St. LOUIS, Inc.</td>
<td>FAA STC</td>
<td>ST00758W</td>
<td>BOMBARDIER CL-600-2B16 (604) BOMBARDIER CL-600-2B16 (601-3R) BOMBARDIER CL-600-2B16 (601-3A)</td>
<td>Cabin Interior Arrangement.</td>
<td>Midcoast Aviation Certification Drawing List 34-269G001, Revision D, dated 18 March 1999, or later approved revision.</td>
<td>This STC must be installed concurrent with an emergency lighting system per EASA STC No. EASA.IM.A.S.02743 and an oxygen system perEASA.IM.A.S.02744 or other approved emergency lighting and oxygen system.</td>
<td>20/03/2008</td>
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<td>EASA.IM.A.S.02747</td>
<td>AMERICAN PROPeller SERVICE</td>
<td>FAA STC</td>
<td>SA5896WM</td>
<td>210L, 210M, 210F, 210G, 210H, 210J, 210K</td>
<td>Installation of a McCauley D33A34C40480*-A6-0 Propeller.</td>
<td>As defined in FAA STC SA5896WM. Prior to installation of this modification, the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Installation in accordance with American Propeller Service, Flight Manual Supplement for SA5896WM, dated 14 January 1993, or later approved revision.</td>
<td>02/10/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.02749</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC</td>
<td>C-LSA08-089D</td>
<td>BOMBARDIER BD-100-1A10 (CHALLE)</td>
<td>AIRCRAFT COMPLETION, 10 PASSSENGERS INTERIOR CONFIGURATION EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA08-089D Issue 1 dated 21st April 2008, in accordance with Aero Consulting Services Ltd. Modification Data Summary F791000 Revision H.</td>
<td>As defined in FAA STC SA5896WM. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>23/04/2008</td>
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<td>EASA.IM.A.S.02750</td>
<td>FIELD AVIATION COMPANY INC.</td>
<td>TCCA STC</td>
<td>SA05-73</td>
<td>BOMBARDIER DHC-8-311</td>
<td>Installation of a Avtec C460-2 or C460-N Three Frequency Emergency Locator Transmitter (ELT).</td>
<td>Configuration A – Ultra Electronics AVC-Active Noise and Vibration Control System and PBMS – Propeller Balancing and Monitoring System. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Maintenance Manual Supplement 202-301</td>
<td>11/06/2008</td>
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<td>EASA.IM.A.S.02751</td>
<td>FIELD AVIATION COMPANY INC.</td>
<td>TCCA STC</td>
<td>SA00-2</td>
<td>BOMBARDIER DHC-8-315 BOMBARDIER DHC-8-311 BOMBARDIER DHC-8-300 SERIES BOMBARDIER DHC-8-300 SERIES</td>
<td>Ultra Electronics Avtec-Active Noise and Vibration Control System and PBMS – Propeller Balancing and Monitoring System. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>Configuration A – Ultra Electronics AVC-Active Noise and Vibration Control System, fitted with a Post 1999/08 TECHNIA INTERIOR (Structural and Electrical, Provisions Only PBMS – Propeller Balancing and Monitoring System. Prior to installation of this modification b...</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST0543AT-D</td>
<td>GULFSTREAM GV-X</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4113.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List, GC41491413, rev. D, dated 19 May 2008, or later approved revisions.</td>
<td>Aircraft Serial Number 4113 only. The following STCs are required part of this STC: ST03027AT-D, ST03047AT-D, ST03032AT-D, ST03026AT-D, ST03031AT-D, ST03024AT-D. It is the installer's responsibility to ensure these STCs are all EASA approved.</td>
<td>Prior to installation of this modification the installer must...</td>
<td>23/05/2008</td>
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<td>EASA.IM.A.S.02757</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST03410AT-D</td>
<td>GULFSTREAM GV-X</td>
<td>Installation of a AR Lavatory Multiple STC</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Index List, GC420000478, Rev N, dated January 29, 2009, or later FAA approved revision.</td>
<td>1. This AR Lavatory Multiple STC, (FAA ref ST03410AT-D), does not approve the entire airplane completion and interior installation and has the following limitations as defined in document GC414660146, &quot;Guidance for Follow on Installation, AR Lavatory Multiple STC&quot;: 1. A complete airplane interior...</td>
<td>08/06/2009</td>
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<td>EASA.IM.A.S.02762</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>FAA STC ST02712CH</td>
<td>BOEING 747-400</td>
<td>Installation of Rosemount Aerospace 8700C2-6 Series Laptop Docking Station (LDS) - Electronic Flight Bag (EFB) Systems Provisions in accordance with Rosemount Aerospace Master Data List, Doc. No. D06402649 Revision C, dated April 3, 2009 or later FAA approved revision.</td>
<td>FAA approved Airplane Flight Manual Supplement, Rosemount Aerospace Doc. No. D06402687, Revision A, approved April 2, 2009 or later FAA approved revision, is required approved on board of the modified aircraft. This installation approval does not include the Electronic Flight Bag computers. The ops...</td>
<td>23/04/2009</td>
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<td>EASA.IM.A.S.02763</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC SA09132AC-D</td>
<td>CESSNA 172N</td>
<td>Installation of S-TEC System 55/55X Two Axis Automatic Flight Guidance System, Model ST-612 in Cessna 172M and 172N.</td>
<td>Bulletin No. 712, Revision 2, dated 28-Feb-2001 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.A.S.02765</td>
<td>PRATT &amp; WHITNEY</td>
<td>FAA TCA Sta 000380, LBA SA 1346</td>
<td>BOMBARDIER B300 B300</td>
<td>Installation of Pratt &amp; Whitney Engine Services, Inc. Twin Engine Aircraft Data Acquisition System ADAS/ADAS+ Upgrade. PN TWINJX-010</td>
<td>07/03/2008</td>
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<td>BOMBARDIER INC.</td>
<td>TCCA STC C-LSA08-145/D ISSUE 1</td>
<td>BOMBARDIER BD-100-1A10 (CHALLENGER)</td>
<td>CUSTOMIZED INTERIOR COMPLETION – AIRCRAFT 200310 passenger seats interior configurationEASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA08-145/D Issue 1 dated 18th August 2008, in accordance with Aero Consulting Services Ltd. Modification Data Summary F827000 Revision 1/2/3</td>
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<td>G&amp;D AERO PRODUCTS INC.</td>
<td>FAA STC SA0763M</td>
<td>BOMBARDIER BEECH 300 BEECH 200</td>
<td>Installation of Acrylic window liners</td>
<td>05/03/2008</td>
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<td>AVIATION ENGINEERING LIMITED</td>
<td>TCCA STC C-LSA08-017/D ISSUE 2</td>
<td>BD-700-1A10</td>
<td>Cabin Electronics Upgrade</td>
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<td>DECA AVIATION ENGINEERING LIMITED</td>
<td>REV 1</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Cabin Interior Reconfiguration</td>
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<td>FIELD AVIATION COMPANY INC.</td>
<td>TCCA STC C-LSA08-033/D</td>
<td>BOMBARDIER CRJ-8-311</td>
<td>VHF Comm Replacement for 8.33 kHz Spacing, ref. TCCA LIST C-LSA08-033/D.</td>
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<td>THUNDER BAY AVIATION Ltd.</td>
<td>TCCA STC- SA04-55</td>
<td>BOMBARDIER DHC-6</td>
<td>Double and Single Utility Seats for DHC-6.</td>
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<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA C- LSA08-062/D ISSUE 1</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Customized Interior Completion - ACSN 9250.</td>
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<td>CHIPPEWA AEROSPACE, Inc.</td>
<td>FAA STC ST03364AT</td>
<td>BOMBARDIER CL-600-2D24 BOMBARDIER CL-600-2D15 BOMBARDIER CL-600-2C10 BOMBARDIER CL-600-2B19</td>
<td>Installs of Protective Covers for the 1K4XD, TCAS and HSTU Processor.</td>
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<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>C-LSA08-065/ D ISSUE 1</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Customized Interior Completion - ACSN 9269</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03546AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4115.</td>
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<td>GKN AEROSPACE CHEM-TRONICS Inc</td>
<td>FAA STC ST01884L-A</td>
<td>AIRBUS A321 AIRBUS A320 WILLIAM BON AIRBUS A319</td>
<td>Installation of ASTECH Engineered Products fan cowl door in accordance with ASTECH’s Master Drawing List V2500 Core Door Assembly No. MDL030-16103, Revision &quot;A&quot;.</td>
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Restricted to ASN 9250 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS 2. INLAN system is compatible only to IEEE 802.11(b). 3. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product 1. The installation of this modification by third parties...

Restricted to ASN 9269 WLAN System is only compatible to IEEE802.11 (b).

Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product 1. The installation of this modification by third parties...

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<td>THUNDER BAY AVIATION Ltd.</td>
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<td>BOMBARDIER COMPLETION CENTRE INC</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
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<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4115.</td>
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<td>AIRBUS A321 AIRBUS A320 WILLIAM BON AIRBUS A319</td>
<td>Installation of ASTECH Engineered Products fan cowl door in accordance with ASTECH’s Master Drawing List V2500 Core Door Assembly No. MDL030-16103, Revision &quot;A&quot;.</td>
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Restricted to ASN 9269 WLAN System is only compatible to IEEE802.11 (b).

Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product 1. The installation of this modification by third parties...

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<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4115.</td>
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<td>FAA STC ST01884L-A</td>
<td>AIRBUS A321 AIRBUS A320 WILLIAM BON AIRBUS A319</td>
<td>Installation of ASTECH Engineered Products fan cowl door in accordance with ASTECH’s Master Drawing List V2500 Core Door Assembly No. MDL030-16103, Revision &quot;A&quot;.</td>
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<td>SIERRA INDUSTRIES Ltd.</td>
<td>FAA STC SA1695WE</td>
<td>CESSNA 172P</td>
<td>Installation of drooped ailerons in flaps down mode, contoured wingleading edges, except for 172M, 172N, and 172P, and wing stallfences.</td>
<td>In accordance with FAA sealed Robertson Aircraft Drawing List No 10.</td>
<td>1. Floatplane Models 172A thru 172K are only eligible with Franklin Engines 6A-350-C1 per STC SA3400NN (215HP) or 6A-350-C2 per STC SA1334RE (180 HP) 2. Fixed tabs as shown on RSTOL Systems, Inc. Drawing 10-100 required on Floatplane Model 172E. The remaining 172 Series Floatplane Models must...</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST02085LA-D</td>
<td>GULFSTREAM Gvi-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial Number 5179.</td>
<td>Install in accordance with Gulfstream Aerospace with Gulfstream Aerospace Report GC52000K555, Rev. N/C and Index List GC52000881 Rev. L, dated 1 May 2008, or later approved revisions. Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M072, Revision N/C, dated 1 May 200...</td>
<td>Aircraft Serial Number 5179 only.</td>
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<td>LEARJET Inc.</td>
<td>FAA STC ST02023 LA-D</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Installation of a Mid-Cabin Interior Modification.</td>
<td>Bombardier Aerospace Master Document List BAE26301021, Revision &quot;F&quot;, dated February 7, 2008 or later approved revision. Installation for Continued Airworthiness (ICA) for the -1 Configuration/Installation are contained in Document Number BAE26301021. ICAa-1, dated January 18, 2008 or later approved re...</td>
<td>A prior or concurrent requirement for the installation of the -1 Configuration is the incorporation of EASA STC No. EASA.IM.A.S.02392 for the interior/Modification Installation.</td>
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<td>07/06/2008</td>
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<td>EASA.IM.A.S.02792</td>
<td>BOMBARDIER INC.</td>
<td>STC C-LSA08-107/D</td>
<td>BOMBARDIER BD-100-1A10 (CHALLENGER 600)</td>
<td>AIRCRAFT COMPLETION: 10 PASSENGERS INTERIOR CONFIGURATION: EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA08-107/D Issue 1 dated 19th March 2008, in accordance with Aero Consulting Services Ltd. Modification Data Summary F786000 Revision N/C.</td>
<td>Transport Canada Limited Supplemental Type Certificate C-LSA08-107/D Issue 1, dated 19th March 2008.</td>
<td>This STC is limited to aircraft Bombardier BD100-1A10 Serial Number 20123.</td>
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<td>BOMBARDIER COMPLETION CENTRE INC</td>
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<td>TC-STA-0113</td>
<td>BOEING 737-500</td>
<td>Customized Interior Completion - ACSN 9262</td>
<td>TC00 STC No. 924-001-13 issue 1 dated 28 October 2004, TC00 STC SOC letter C-09-014 dated 28 May 2008, Aero Consulting Services Configuration Deviation List D064000 Revision B dated 09 May 2008 Bombardier Airplane Flight Manual Supplement G-FM25004000 Revision NCI. Instructions for Continued Airworthiness G-</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>09/06/2008</td>
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<td>BOMBARDIER COMPLETION CENTRE INC</td>
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<td>BOMBDIER-BD-700-10</td>
<td>Customized Interior Completion - ACSN 9273</td>
<td>TC00 STC No. 945-60 issue 1 dated 14 June 2005, TC00 STC SOC letter C-08-026 dated 02 September 2008, Aero Consulting Services Modification Data Summary E025000 Revision AW dated 29 August 2008 Bombardier Airplane Flight Manual Supplement G-FM25004000 Revision NCI. Instructions for Continued Airworthiness G-</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the aircraft.</td>
<td>10/06/2008</td>
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<td>EASA.IM.A.S.02796</td>
<td>TDG AEROSPACE INC</td>
<td>-</td>
<td>FAA STC ST00010A</td>
<td>BOEING 737-900</td>
<td>Universal Fault Interrupter Installation Kit for B737NG.</td>
<td>FAA approved Airplane Flight Manual Supplement (Ref. TDG document 26-3202-0003), Aircraft Maintenance Manual (Ref. TDG document 26-3202-0001), Aircraft Wiring Diagrams Manual (Ref. TDG document 26-3202-0005), Aircraft Illustrated Parts Catalogue (Ref. TDG document 26-3202-0006) or later FAA approved...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the aircraft.</td>
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<td>PANASONIC AVIONICS CORPORATION</td>
<td>-</td>
<td>BOEING 757-200 SERIES</td>
<td>Installation of a Matsushita Avionics Audio System on a Boeing 757-200 Aircraft.</td>
<td>EJ. Mlynarczyk Drawing List DL10053. Revision H. 28/09/1996</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the aircraft.</td>
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<td>TTF AEROSPACE, LLC</td>
<td>FAA STC ST01966E</td>
<td>BOMBARDIER DHC-8-300 SERIES</td>
<td>Installation of a Reinforced Flight Deck Door.</td>
<td>TTF Aerospace, LLC Master Data List (MDL), TTF-MDL-07011, Rev. J, dated 05.10.2007 Or later approved revisions of the above document approved by EASA in accordance with EASA ED Decision 2004/04/CF or subsequent revisions of this decision.</td>
<td>TTF Aerospace, LLC Master Data List (MDL), TTF-MDL-07011, Rev. J, dated 05.10.2007 Or later approved revisions of the above document approved by EASA in accordance with EASA ED Decision 2004/04/CF or subsequent revisions of this decision.</td>
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<td>EASA.IM.A.S.02800</td>
<td>JAZZ AVIATION LP</td>
<td>TC STC SA03-70</td>
<td>BOMBARDIER DHC-8-200 SERIES, BOMBARDIER DHC-8-100 SERIES</td>
<td>Cockpit Door Electronic Strike System.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. This AFM Supplement is applicable only when used in conj...</td>
<td>TTF Aerospace, LLC Master Data List (MDL), TTF-MDL-07011, Rev. J, dated 05.10.2007 Or later approved revisions of the above document approved by EASA in accordance with EASA ED Decision 2004/04/CF or subsequent revisions of this decision.</td>
<td>30/06/2008</td>
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<td>FLYING INVESTMENTS, LLC</td>
<td>FAA STC ST01348W</td>
<td>MYSTERE FALCON 20- F5, MYSTERE FALCON 20- E5, MYSTERE FALCON 20- D5, MYSTERE FALCON 20- C5, FAN JET FALCON SERIES G, FAN JET FALCON SERIES E, FAN JET FALCON SERIES D</td>
<td>Installation of an Innovative Solutions &amp; Supports (IS&amp;S) Dual Air Data Display Units (ADDU), an Analog Interface Unit or Air Data Sensor and Goodrich Dual Pitot-Static Probes, in accordance with Flying Investments LLC Master Data List GEN-031217-001, Revision K, dated 04th December 2007 or later EASA approved Revisions.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person...</td>
<td>Flying Investments LLC Master Data List GEN-031217-001, Revision K, dated 04th December 2007 or later EASA approved Revisions.</td>
<td>22/07/2008</td>
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<td>EASA.IM.A.S.02802</td>
<td>FLYING INVESTMENTS, LLC</td>
<td>FAA STC ST05710E</td>
<td>MYSTERE FALCON 20- F5, MYSTERE FALCON 20- E5, MYSTERE FALCON 20- D5, MYSTERE FALCON 20- C5, FAN JET FALCON SERIES G, FAN JET FALCON SERIES E, FAN JET FALCON SERIES D</td>
<td>Installation of a Master Data List (MDL), TTF-MDL-07011, Rev. J, dated 05.10.2007 Or later approved revisions of the above document approved by EASA in accordance with EASA ED Decision 2004/04/CF or subsequent revisions of this decision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person...</td>
<td>Master Data List GEN-031217-001, Revision K, dated 04th December 2007 or later EASA approved Revisions.</td>
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<td>TCCA C-LSA08-081/D</td>
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<td>Interior Completion, BD-700-1A10 serial number 9252</td>
<td>1. TCCA STC No. C-LSA08-081/D issue 1 dated 22 October 2008.2 TCCA Soc reference C-08-2001 dated 30 October 2008.3. Aero Consulting Services Modification Data Summary FT45000 Revision N/C or later approved revision A Bombardier Flight Manual Supplement G-FM250001172 Revision NC or later.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third party.</td>
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<td>TCCA STC C-LSA08-082/D</td>
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<td>Interior Completion, BD-700-1A10 serial number 9256.</td>
<td>1. TCCA STC No. C-LSA08-082/D, issue 1 dated 10 October 2008.2 TCCA Soc Reference C-08-2002 dated 23 October 2008.3. Aero Consulting Services Modification Data Summary FT46900 Revision N/C4 Bombardier Flight Manual Supplement G-FM250001189 Revision N/C5. GC92560000-IFCA Instructions.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third party.</td>
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<td>EASA.IM.A.S.02809</td>
<td>SIERRA INDUSTRIES Ltd.</td>
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<td>FAA STC SA4375W</td>
<td>CESSNA 550 (CITATION II) CESSNA 550 CITATION II</td>
<td>Installation of radome quick release mechanism FAA STC SA4375W.</td>
<td>Sierra Industries Inc. Drawing List No. S116-000, Rev. B, dated 13 July 1992 or later approved revisions of this document approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision). This STC extends the applicability of LBA STC Nr. TA009, dated 26 August 2009.</td>
<td>Sierra Industries Inc. Drawing List No. S116-000, Rev. B, dated 13 July 1992 or later approved revisions of this document approved by EASA in accordance with EASA ED Decision 2004/02/CF (or subsequent revisions of this decision). This STC extends the applicability of LBA STC Nr. TA009, dated 26 August 2009.</td>
<td>28/03/2008</td>
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<td>EASA.IM.A.S.02812</td>
<td>GENERAL AEROSPACE INC</td>
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<td>FAA STC ST03486AT</td>
<td>BOEING 737-500 BOEING 737-400 BOEING 737-300 BOEING 737-200</td>
<td>Installation of a Honeywell ARINC 717 Flight Data Acquisition Management System (FDAMS).</td>
<td>FAA letter of compliance with EASA Certification Basis dated 12 March 2008 Master Drawing List No. GA37-0FDAUADB017, Revision A, dated 24 April 2007 or later revision of the above-listed documents approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>FAA letter of compliance with EASA Certification Basis dated 12 March 2008 Master Drawing List No. GA37-0FDAUADB017, Revision A, dated 24 April 2007 or later revision of the above-listed documents approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>19/03/2008</td>
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<td>EASA.IM.A.S.02813</td>
<td>CAV AEROSPACE Inc.</td>
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<td>FAA STC SA1435WI</td>
<td>LC41-550FG (C400)</td>
<td>Installation of TKS Ic Protection System Components. Model/LC41-550FG (C400)</td>
<td>According to ASAT or CAV Master Drawing List No. 12200-DL, Rev. 2 or later FAA-approved RevisionsTo be operated and maintained in accordance with FAA approved ASAT or CAV AFM &amp; Document No. 12200-FMS, original issue, or later FAA-approved Revisions</td>
<td>According to ASAT or CAV Master Drawing List No. 12200-DL, Rev. 2 or later FAA-approved RevisionsTo be operated and maintained in accordance with FAA approved ASAT or CAV AFM &amp; Document No. 12200-FMS, original issue, or later FAA-approved Revisions</td>
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<td>EASA.IM.A.S.02815</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC SAB228W-D</td>
<td>FALCON 2000</td>
<td>Installation of the Airborne Flight Information System.</td>
<td>Dassault Falcon Jet Drawing List Report No. F2M-1055, Revision B dated 30th May 1999 and later revisions approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>See those of original FAA STC SAB228W-D Revision 2 dated 21st May 1999 or later revisions approved by EASA in accordance with EASAED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>20/05/2008</td>
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<td>EASA.IM.A.S.02816</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA C-LSA08-143D</td>
<td>GIV</td>
<td>Installation of Interior Modifications.</td>
<td>Installation in accordance with Aero Consulting Services Ltd Modification Data Summary F610000 Rev NC or later approved revisions. Maintenance in accordance with Aero Consulting Services Ltd Mandatory Maintenance Manual Supplement F610071 Rev NC and F610070 Instructions for Continued Airworthiness...</td>
<td>Aircraft Serial Number 1109 Only Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/06/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.02817</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA C-LSA08-151D</td>
<td>GIV</td>
<td>Installation of Rockwell Collins Radio Tuning Units RTU-4210 and Rockwell Collins TDR-400 (EL/ESHS) Transponder.</td>
<td>Installation in accordance with Aero Consulting Services Ltd Modification Data Summary F773000 Rev NC or later approved revisions. Maintenance in accordance with Aero Consulting Services Ltd Mandatory Maintenance Manual Supplement F773073 Rev NC and F773070 Instructions for Continued Airworthiness...</td>
<td>Application to Aircraft Serial Number 1109 only. The operator is responsible for the exclusion of a 24 month check in their maintenance schedule of the Mode S transponder system. Townsure correct operation of the 24BIT address and the required DAPs IAW ACS MM 34-53-00. For limitations refer to the Airworthi...</td>
<td>11/06/2008</td>
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<td>EASA.IM.A.S.02818</td>
<td>BJAC BUSINESS JET AIRCRAFTCOMPL ETIONS</td>
<td>FAA SA07-129 ISSUE 1</td>
<td>BD-705-1A10, BD-700-1A11</td>
<td>Goodrich Class III Electronic Flight Bag.</td>
<td>TCCA STC No. S407-129 Issue 1 dated 20 November 2007.TCCA SOC 5010-020-0020 dated 08 October 2008.Avironic Design Services Ltd. Master Drawing List MDL9079 Revision N/C dated 31st November 2007, or later approved revision BJAC Services Flight Manual Supplement FMB0079-1 Rev N/C, approved 20 No...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>14/10/2008</td>
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<td>ANORA INTERNATIONAL LLC</td>
<td>ST02388CH</td>
<td>STC ref.</td>
<td>FAA STC ST02388CH</td>
<td>This STC may not be installed after 7th June 2010, unless: 1. Prior to installation of this modification, the installer must determine that the installation of this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person...</td>
<td>01/07/2009</td>
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<td>ELLIOTT AVIATION</td>
<td>ST02548CH</td>
<td>EASA Validation of FAA approved STC</td>
<td>FAA STC ST02548CH</td>
<td>Installation of a Universal Satellite Telephone System</td>
<td>26/05/2008</td>
<td>Active</td>
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<td>ELLIOTT AVIATION</td>
<td>ST02547CH</td>
<td>EASA Validation of FAA approved STC</td>
<td>FAA STC ST02547CH</td>
<td>Installation in accordance with Elliott Aviation Master Data List, Report No. MDL-2636-114 Revision D, dated 23rd April 2008, or later revisions of this document approved by EASA in accordance with EASA ED Decision 2004/04/CF (i.e. a specific FAA AFM Supplement Elliott Aviation report No. FMS-2637-11; Revision IR, approved 14th March 2008 is required on board the aircraft. Modified aircraft must be maintained in accordance with EASA in accordance with FAA AFM Supp...</td>
<td>26/05/2008</td>
<td>Active</td>
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<td>AIRWORKS</td>
<td>ST02150LA</td>
<td>EASA Validation of FAA approved STC</td>
<td>FAA STC ST02150LA</td>
<td>As per FAA approved Air Works Master Data List, document MDL23019, rev. B, dated 20 June 2008. This STC may not be installed after 7th June 2010, unless: 1. Prior to installation of this modification, the installer must determine that the installation of this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person...</td>
<td>10/07/2008</td>
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<td>PREMIER AIR CENTER, LLC</td>
<td>ST02386CH</td>
<td>EASA Validation of FAA approved STC</td>
<td>FAA STC ST02386CH</td>
<td>Installation of a Rockwell Collins IFIS-5000 Integrated Flight Information System with display on the existing Rockwell Collins Proline21 Continuum Electronic Flight Instrument System (EFIS) / EASA Validation of FAA STC ST02386CH</td>
<td>01/07/2009</td>
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**List of EASA Supplemental Type Certificates**

1. Prior to installation of this modification, the installer must determine that the installation of this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person...
<table>
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<tr>
<th>EASA STC no.</th>
<th>STC Holder</th>
<th>FAA STC</th>
<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
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<tr>
<td>EASA.IM.A.S.02834</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01730LA-D</td>
<td>BOMBARDIER CL-600-2B16 (604)</td>
<td>FAA STC ST01730LA-D, Installation of an Aircell Axxess II Iridium Satcom System.</td>
<td>FAA STC ST01730LA-D dated 15 August 2006.Bombardier Aerospace Master Document List BA683017007, Revision B, dated August 19, 2006, or later Approved revision.The Maintenance Instructions for Continued Airworthiness (IFCA) for the -1 Configuration / Installation are contained in Document Number BA683.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/06/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.02836</td>
<td>DECA AVIATION ENGINEERING LIMITED</td>
<td>TC STC O-LSA08-069/D</td>
<td>BOMBARDIER DHC-6-300</td>
<td>Radar System Installation.</td>
<td>Installation in accordance with DECA Aviation Engineering Ltd Modification Summary MS 07099 Rev 1 or later Transport Canada approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>12/11/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.02837</td>
<td>LIFEPART, LLC</td>
<td>FAA STC ST02119LA</td>
<td>CESSNA 750 CITATION X</td>
<td>Emergency Medical System</td>
<td>Master Drawing List, MDL-833-06, Revision DAirplane Flight Manual EGR 833-15, Revision AInstructions for Continued Airworthiness, EGR 833-12-A</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>30/07/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.02838</td>
<td>AVIATION PARTNERS BOEINGWINGLET COMPANY, LLC</td>
<td>FAA ST01219SE</td>
<td>737-300, 737-500</td>
<td>Upgrade Super King Air 200, A200 and B200 Series King Airs to Pratt and Whitney Canada PT6A-52 engines.</td>
<td>Installation of Pratt &amp; Whitney Pratt &amp; Whitney Canada PT6A-52 in accordance with Blackhawk Modifications Inc. Master Drawing List No. 05210050, Rev. IR, dated 15. Nov. 2007 or later approved revision.Operation must be inaccordance with Blackhawk Modifications Inc. Aircraft Flight...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/07/2009</td>
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<td>EASA.IM.A.S.02839</td>
<td>BLACKHAWK MODIFICATIONS Inc.</td>
<td>FAA SA10248C</td>
<td>B. B200T B. 200, 200C, 200CT, 200T A. B200, B200C, B200CT</td>
<td>Forward LED Navigation Lighting System Installation</td>
<td>FAA STC ST10742SCC Aircraft Systems and Manufacturing Top Drawing List, ref. 5655-0001-00, Rev. C dated 19.07.2007 Instruction for Continued Airworthiness, ref. R-398-13, Rev. A dated 15.11.2007</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/06/2008</td>
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### List of EASA Supplemental Type Certificates

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<th>Models</th>
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<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<tr>
<td>EASA.IM.A.S.02843</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST10206SC-D</td>
<td>GULFSTREAM 200</td>
<td>Installation of a 3rd FMS 6100 Flight Management System i.a w. Index List GA21018500; Rev C or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004044CF (or subsequent revisions of this decision).</td>
<td>AFMS GA21018500 Rev - or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004044CF (or subsequent revisions of this decision).</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>28/05/2008</td>
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<td>EASA.IM.A.S.02844</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST10204SC-D</td>
<td>GULFSTREAM 200</td>
<td>Installation of a Pedestal Extension in a Gulfstream G200.</td>
<td>Index List GA2200408000, Rev A or later revision of the above document approved by EASA in accordance with EASA ED Decision 2004044CF (or subsequent revisions of this decision).</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.A.S.02845</td>
<td>LIFEPOR, LLC</td>
<td>FAA STC SA00059LA</td>
<td>HAWKER HS 125-700A</td>
<td>Installation of a Patient Loading and Utility System.</td>
<td>Installation in accordance with LifePort Inc. Master Drawing List as defined in AML No. EASA.IM.A.S.02845 to this STC. Dated 30 July 2008 or later approved revisions.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>30/07/2008</td>
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<td>EASA.IM.A.S.02846</td>
<td>STG AEROSPACE Inc.AKA SAFTGLO, Inc.</td>
<td>FAA STC ST01829AT</td>
<td>BOEING 737-700</td>
<td>Installation of the SAFTGLO Non-electrical PhotoluminescentEscape Path Marking System to include all 737 models.</td>
<td>The photo luminescent Floor Proximity Emergency Escape Path Marking System is only eligible for installation on aircraft model numbers with interior arrangements approved in the SafTGlo MDL No. E74-2, Rev. I.R., dated 9 February 1999*. <em>See DAS/ SafTGlo MDL Report Number E74-2, revision G, dated 13 June 2002</em> for installations occurring after 1 July 2002 *or any later revision of the above listed documents approved in accordance with...</td>
<td>14/04/2008</td>
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<td>EASA.IM.A.S.02853</td>
<td>EJM AEROSPACE SERVICES</td>
<td>FAA STC ST02534AT</td>
<td>BOEING 747-400</td>
<td>Installation of a Matsushita Avionics Systems Corporation System 3000 In-Flight Entertainment System on a Singapore Airlines Boeing 747-400 in accordance with Master Drawing List MDL 1021, Rev. H, dated 26/04/2003, or later FAA approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data. Compatibility with other aircaft/engine configurations shall be determined by the installer.</td>
<td>18/08/2008</td>
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<td>EASA.IM.A.S.02854</td>
<td>LIFEPOR, LLC</td>
<td>FAA STC ST00194SE</td>
<td>CESSNA 050</td>
<td>Installation of a Pedestal Extension in a Gulfstream G200.</td>
<td>Installation in accordance with LifePort Inc. Manual Supplement for LifePort/PLS System No. 39 as defined in AML No. EASA.IM.A.S.02845 to this STC. Dated 30 July 2008 or later approved revisions.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.S.02855</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA C-LSA08-097/D ISSUE 1</td>
<td>Installation of Club Seating</td>
<td>TCCA STC No. CL-LSA08-097/D, Issue 1, dated 20 March 2008 Airlfe Aeronautical Limited (AIL) Master Data List MSL418601 Revision IR dated 2008 March 18 AAIL Instructions for Continued Airworthiness document IC411601, Revision IR dated 2008 March 18 or later approved revision.</td>
<td>1. Restricted to ASN 54642. Limitation: This approval is limited to 1,600 flight cycles or 12 months from the approval date, whichever comes first until the Airworthiness Limitations subdocument BA63030116-AVL-1 of Bombardier IFCA BAS63030116-IFCA-1 is approved. 3. Prior to installation of this modif...</td>
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<td>EASA.IM.S.02856</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA C-LSA07-418/D ISSUE 1</td>
<td>Installation of EMS High Speed Data System</td>
<td>Transport Canada STC No C-LSA07-4180 Issue 1 dated 31 December 2007 Bombardier Master Data List BASS3010116, Revision A dated 14 March 2008 or later approved revision .Bombardier Instructions for Continued Airworthiness (IFCA) document BASS3010116-IFCA-1, Revision A or later accepted (AWL subdoc...</td>
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<td>EASA.IM.S.02858</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC ST04039NY-0</td>
<td>Installation of Saat In-Flight Entertainment Components and Software update in accordance with Delta Engineering Master Data List 0107-06195-2172 Revision A, dated 4 April 2008. Validation of FAA STC ST04039NY-0.</td>
<td>0107-06195-2172 Rev. A Master Data List for Panasonic Avionics IF System 3000: 44-35-72 Aircraft Maintenance Manual Supplement B747-400/0070-06201-2172 Rev. A. Instruction for Continued Airworthiness for System 3000 IF installed in B747-400</td>
<td>Prior to installation of this modification the installer must determine that the interaction between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person...</td>
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<td>EASA.IM.S.02859</td>
<td>BASIC AIRCRAFT PRODUCTS, INC.</td>
<td>FAA SA02747AT</td>
<td>182</td>
<td>As per FAA STC SA02747AT limitations.</td>
<td>29/08/2008</td>
<td>Active</td>
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<td>EASA.IM.S.02860</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST02533CH-D</td>
<td>Aircraft Serial No 5186 completion and cabin interior installation.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2004832 Original Revision A, dated July 8, 2008. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 2004832, Original Release, dated July 15&quot; for later revisions of the above listed document...</td>
<td>1. GV-SP Serial No. 5186 Only. Prior to installation of this modification the installer must determine that the interaction between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation...</td>
<td>29/07/2008</td>
<td>Active</td>
<td>1016498</td>
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<tr>
<td>EASA.IM.S.02861</td>
<td>HOLLINGSEAD INTERNATIONAL</td>
<td>FAA STC 01863LA BOEING 767-300 BOEING 767-200</td>
<td>Installation of mounting provisions and electrical provisions for electronic flight bags</td>
<td>Installation in compliance with Hollinghead International Master Data List 2813-01 Revision D, February 20, 2007</td>
<td>Prior to installation of this modification the installer must determine that the interaction between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/06/2008</td>
<td>Active</td>
<td>1016499</td>
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<td>STC Holder</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>ASTRONICS ADVANCED ELECTRONICS SYSTEMS</td>
<td>FAA ST02515AT</td>
<td>747-400</td>
<td>Installation of a General Dynamics EmPower AC In-Seat powerSupply System on a Boeing 747-400 type aircraft in accordance with Master Drawing list MLD1091-3, Rev. C, dated August 6, 2002, or later FAA approved revision</td>
<td>FAA approved Airplane Flight Manual Supplement Document Number D1091-517-8, dated June 6, 2002 or later approved revision</td>
<td>Prior to installation of this modification the installer must determine the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>23/07/2008</td>
<td>Active</td>
<td>10/12302</td>
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<tr>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA C-LSA08-237/D</td>
<td>BD-100-1A10</td>
<td>CUSTOMIZED INTERIOR COMPLETION - AIRCRAFT 2012120 passengers interior configuration EASA Validation of Transport Canada Limited Supplemental Type Certificate C-LSA08-237/D Issue 1 dated 24th September 2009, in accordance with Aero Consulting Services Ltd. Modification Data Summary F925000 Revision...</td>
<td>Transport Canada Limited Supplemental Type Certificate C-LSA08-237/D Issue 1, dated 24th September 2009. Transport Canada Approved Aero Consulting Services Ltd. Modification Data Summary ref. F925000 dated 24th September 2009. Revision N/C or later Transport Canada approved revision.</td>
<td>1 Prior to installation of this modification the installer must ensure the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third...</td>
<td>09/10/2008</td>
<td>Active</td>
<td>10/15202</td>
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<tr>
<td>LIVETV, L.L.C.</td>
<td>FAA STC ST03177AT</td>
<td>EMBRAER ERJ 190-100 ECJ</td>
<td>FAA STC ST03177AT: Installation of LiveTV LTV 2000 In-Flight Entertainment System</td>
<td>- EML 1191/TK010-MDL, Revision E, dated June 14, 2006 - J1901TR0010-AFM dtd. October 18, 2005 - Instructions for Continued Airworthiness J1901TR0010CA Rev. IR dated August 3,2005 (*) or later approved revision (see EASA EC Decision 2607/2010C or later revisions of this Decision)</td>
<td>Installation based on EMBRAER provisions</td>
<td>05/06/2008</td>
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<td>10/15203</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST00292LA-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5191.</td>
<td>Install in accordance with Gulfstream Aerospace Report GC52000K574, Rev. N/C and Index List GC520008319 Rev. L, dated 28 September 2008, or later approved revisions. Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M078, Revision A, dated 2 September 2008.</td>
<td>1. Aircraft Serial Number 5191 only. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation...</td>
<td>11/06/2008</td>
<td>Active</td>
<td>10/15204</td>
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<td>CHPPEWA AEROSPACE, Inc.</td>
<td>FAA STC ST03573AT</td>
<td>BOMBARDIER CL-600-2D4</td>
<td>BOMBARDIER CL-600-2D1S BOMBARDIER CL-600-2D10 BOMBARDIER CL-600-2B19</td>
<td>Installs the electrical provisions for Goodrich #735/A2 Class 2 Elec. Flt. Bag</td>
<td>FAA Statement of Compliance dated 29 May 2008 FAA STC No. ST03573AT dated 25 April 2008 Chippewa Aerospace MDL No. 601-620-01-0001 revision IR dated 29 January 2008 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. This modification may not be installed after June 7, 2011.</td>
<td>24/06/2008</td>
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<td>10/15205</td>
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<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC ST01883CH</td>
<td>747-300, 747SR 747-200B, 747-200C, 747-200F</td>
<td>Installation of Universal Avionics Systems Corporation Multi-FunctionDisplay in accordance with Electronic Cable Specialists Master Data List ECS-200063 Rev. 0 dated April 18, 2009 or later approved revision.</td>
<td>FAA approved Airplane Flight Manual supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-200063 Rev. G, dated April 18, 2009 or later approved revision, is required on board the aircraft.</td>
<td>1. Airplane Flight Manual supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-200063 Rev. G, dated April 18, 2009 or later approved revision, is required on board the aircraft. Prior to installation of this modification the installer must determine that the...</td>
<td>02/06/2009</td>
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<td>10/243003</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
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<td>Limitations/Conditions</td>
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<td>EASA.IM.A.S.02873</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC ST02873CH</td>
<td>BOEING 747-500F, BOEING 747-600F, BOEING 747-700C, BOEING 747-800</td>
<td>Installation of Elementary or Enhanced Surveillance Mode S Transponder &amp; Electronic Cable Specialists Master Data List EICS-202128, rev. AB, 7 January 2008 or later approved revision.</td>
<td>Electron Cables &amp; Specialised Afm supplement as listed in Appendix A of Electronic Cable Specialists Master Data List EICS-202128, rev. AB, 7 January 2008 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/10/2008</td>
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<td>1018507</td>
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<td>EASA.IM.A.S.02875</td>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD.</td>
<td>TCCA IM.A.S.02875</td>
<td>BOMBARDIER CL-600-2B16</td>
<td>Cabin Finishing Rubbish on aircraft S/N 5442.</td>
<td>1. TCIA STC No. Q-LSA08-021/D dated 11 June 2008. 2. (Innotech Aviation Master Document List MOL128103002 revision B dated 11 June 2008.</td>
<td>1. Restricted to ASIN 5442.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>06/10/2008</td>
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<td>EASA.IM.A.S.02876</td>
<td>L3 COMMUNICATIONS S.A. AVIATION SYSTEMS</td>
<td>FAA STC ST03560AT</td>
<td>DC-9-81, DC-9-82, DC-9-83, DC-9-87</td>
<td>Installation of L3 Communications Avionics Recorder Micro Quick Access Recorder (uQAR).</td>
<td>Master Drawing List No. 905-E4310-29 Rev. D dated 28.02.2008 Instructions for continued Airworthiness No. 905-E4311-20 Rev. A Dated 25.10.2007.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/07/2008</td>
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<td>1018509</td>
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<td>EASA.IM.A.S.02877</td>
<td>AIRCRAFT ENGINEERING &amp; INSTALLATION SERVICES Inc.</td>
<td>FAA STC ST0574AT</td>
<td>767-200</td>
<td>Installation of the KAIHAAQ 408 ATP Emergency Locator Transmitter</td>
<td>Aircraft Engineering and Installation Services Master Drawing List 234-04-0025, Revision B, 11th October 2004</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/06/2008</td>
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<td>1018510</td>
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<td>EASA.IM.A.S.02878</td>
<td>ROCKY MOUNTAIN AIRCRAFT</td>
<td>TCCA STC SA08-38</td>
<td>BOMBARDIER DHC-6 SERIES 200, BOMBARDIER DHC-6-300, BOMBARDIER DHC-6-100</td>
<td>Installation of Pitot Heat Indication system.</td>
<td>Installation in accordance with Transport Canada STC SA08-38, Issue 1, April 07, 2008, or later Transport Canada approved revision. The applicable manuals are as follows: Airplane Flight Manual Supplement – TCCA approved FMS-1270-01, Issue 1, dated 29 February 2008, or later Transport Canada approved...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/11/2008</td>
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<td>EASA.IM.A.S.02879</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0524CH-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Aircraft completion and cabin interior installation Serial No 4119.</td>
<td>1. Installation in accordance with Gulfstream Aerospace Corporation STC SA08-38, Issue 1, April 07, 2008, or later approved revision. The applicable manuals are as follows: Aircraft Flight Manual Supplement – Gulfstream SA08-38, Issue 1, dated 29 February 2008, or later Transport Canada approved...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/08/2008</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>EASA.IM.A.S.02882</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0283CH-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Aircraft Serial No 5197 completion and cabin interior installation.</td>
<td>1. Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2046103, Revision A, dated September 23, 2008, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 204M303, Original Release, dated September 29, 20...</td>
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<td>EASA.IM.A.S.02884</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03573AT</td>
<td>GULFSTREAM 200</td>
<td>Installation of an Auxiliary Power Unit (APU) Surge Control/Valve Modification.</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Index List, CEB24428000, rev D, dated April 3, 2008, or later approved revisions. Operation in accordance with Civil Airworthiness Authority of Israel (CAAI) and FAA Approved Gulfstream 200 Flight Manual Section 1 Limitations, APU Limitations...</td>
<td>02/07/2008</td>
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<td>EASA.IM.A.S.02886</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0283CH-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Aircraft Serial No 5189 completion and cabin interior installation.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2046096, Revision A, dated August 26, 2008, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 204M331, Original Release, dated August 26, 2008, or later...</td>
<td>17/09/2008</td>
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<td>EASA.IM.A.S.02887</td>
<td>HARTZELL SERVICE CENTER</td>
<td>FAA STC SA2073CH</td>
<td>CESSNA R172K</td>
<td>Installation of a Hartzell Propeller Inc. 75” Diameter 3-Blade Propeller model (PHC-G3YF-1RF/F7392 with C-4582-P) spinner on a Cessna R172K (Hawk XP) Landplane aircraft.</td>
<td>STC No. SA02073CH, dated December 3, 2004. Hartzell Document #R172K-NST  &quot;Instructions for Installation of a Hartzell (PHC-G3YF-1RF/F7392 propeller with C-4582-P) spinner on a Cessna R172K (Hawk XP) Landplane aircraft. The STC is compatible with STC SA1437CE that installs the TCM IO-360-K, or –KB, engine as modified by STC SE1436CE. The re-rated engine is designated IO-360-KEC, or –KBC, which increases the power rating to 210hp@2800 rpm for five minutes takeoff and 195hp@2600 rpm maximum continuous...</td>
<td>05/08/2008</td>
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List of EASA Supplemental Type Certificates
EASA STC no.

STC Holder

Rev.
level

Foreign/NAA
STC ref.

Models

Description

Certification basis

Associated Technical
Documentation

easa.europa.eu
Limitations/Conditions

Date of
Issue

Status

System
Record

EASA.IM.A.S.02892

GULFSTREAM
SERVICES
CORPORATION

FAA STC
ST03565AT-D

GULFSTREAM GIV-X

Installation of Custom Cabin Interior and
Aircraft Completion for Aircraft Serial
4122.

Install in accordance with
Gulfstream Aerospace
CorporationMaster Index
List, GC414914122 Rev. F,
dated August 1, 2008or
later approved revisions.

Aircraft Serial Number 4122
only.The following STC's are a
required part of this
STC:ST03027ATDST02987AT-DST03047ATDST02980AT-DST03290ATDST03532AT-DST03238ATDST02979AT-DIt is the
installer's responsability to
ensure these STC's are
allEASA approved.

06/08/2008

Active

10016520

EASA.IM.A.S.02893

GULFSTREAM
SERVICES
CORPORATION

FAA STC
ST03449AT-D

GULFSTREAM GV-SP

Installation of Custom Cabin Interior and
Aircraft Completion for Aircraft Serial
5155.

Installation in accordance
with Gulfstream Aerospace
Corporation Master Index
List GC514915155, Rev. F,
dated September25, 2007
orlater FAA approved
revision.

Aircraft Serial Number 5155
only:the following STC's are a
required part of this
STC:ST02796ATDST02697AT-DST02715ATDST03290AT-DST01775LADST02696AT-DST00892LAIn
addition, this STC must be
installed in conjunction with
Cabin Modification STC
ST02089LA-D to fully meet
EASA Requirements.It is the
ins...

05/08/2008

Active

10016521

EASA.IM.A.S.02895

FLYHT
AEROSPACE
SOLUTIONS Ltd.

TCCA STC
SA07-124

BOEING 767-300
BOEING 767-200

AFIRS 220 Automated Flight Information
Reporting System installation with
Optional Global Voice and Global
Satellite Communication

AMS Master Document
List, Ref. 121-0039 Rev. B,
Dated 3 July 2008.AMS
Installation Instructions,
Ref. 151-0030 Rev. A,
Dated 12 November 2007

Prior to installation of this
modification the installer must
determine that the
interrelationship between this
modification and any other
previously installed modification
will introduce no adverse effect
upon the airworthiness of the
product.The installation of this
modification by third persons i...

19/08/2008

Active

10016522

EASA.IM.A.S.02897

DASSAULT
FALCON JET
CORP.(SERVICE
CENTER)

FAA STC
SA7049SW-D

MYSTERE FALCON 900

Installation of Electrical FeedThru in
Pressure Vessel.

FAA STC: SA7049SW-D,
Revision 10, dated 05 Aug
1994.DrawingList No.:
F90-01048, Revision N,

Prior to installation of this
modification the installer must
determine that the
interrelationship between this
modification and any other
previously installed modification
will introduce no adverse effect
upon the airworthiness of the
product.

23/06/2008

Active

10016524

EASA.IM.A.S.02898

DASSAULT
FALCON JET
CORP.(SERVICE
CENTER)

FAA STC
SA7056SW-D

MYSTERE FALCON 900

Installation of Automatic Cabin Electrical
Load Shed System.

FAA STC: SA7056SW-D,
Revision 6, dated 17 Dec
2000.Drawing List No.:
F90-01012, Revision K,
dated 12 Dec 2000.For
501 Installation -AFM
Supplement No.: 35888-1,
Revision B, dated06 Jan
1988.For 503 or 505
Installation -AFM
Supplement No.: 35888-2,
Revision A, dated 17 Dec
2000.

Prior to installation of this
modification the installer must
determine that the
interrelationship between this
modification and any other
previously installed modification
will introduce no adverse effect
upon the airworthiness of the
product.

23/06/2008

Active

10016525

EASA.IM.A.S.02899

DASSAULT
FALCON JET
CORP.(SERVICE
CENTER)

REV 1

FAA STC
SA7060SW-D

MYSTERE FALCON 900

Installation of Basic Interior Furnishings.

FAA STC: SA7060SW-D,
Revision 68, dated 07 Dec
1996Drawing List No.:
F90-01007, Revision F,
dated 06 Dec
1996Maintenance of
Pocket Door Assembly:

Prior to installation of this
modification the installer must
determine that the
interrelationship between this
modification and any other
previously installed modification
will introduce no adverse effect
upon the airworthiness of the
product.

23/06/2008

Active

10016526

EASA.IM.A.S.02900

DASSAULT
FALCON JET
CORP.(SERVICE
CENTER)

REV 1

FAA STC
SA7062SW-D

MYSTERE FALCON 900

Installation of Basic Oxygen System.

FAA STC: SA7062SW-D,
Revision 4, dated 07 Dec
1996Drawing List No.:
F90-01009, Revision R,
dated 04 Dec 1996.AFM
Supplement No.: 36103-1,

Prior to installation of this
modification the installer must
determine that the
interrelationship between this
modification and any other
previously installed modification
will introduce no adverse effect
upon the airworthiness of the
product.

23/06/2008

Active

10016527

03/07/2024

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An agency of the European Union


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<th>Foreign/NAA STC ref.</th>
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<th>Certification Basis</th>
<th>Limitations/Conditions</th>
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<td>EASA:IMA.S.02901</td>
<td>DASSAULT FALCON JET CORP. (SERVICE CENTER)</td>
<td>REV 1</td>
<td>FAA STC SAT085SW-D</td>
<td>MYSTERE FALCON 900</td>
<td>Installation of Avionics Rack.</td>
<td>FAA STC: ST09309SC-D, Revision 13, dated 19 Oct 2000. Drawing List No.: F90-0102, Revision AC, dated 21 July 1994.</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA:IMA.S.02902</td>
<td>DASSAULT FALCON JET CORP. (SERVICE CENTER)</td>
<td>REV 1</td>
<td>FAA STC SAT123SW-D</td>
<td>MYSTERE FALCON 900</td>
<td>Installation of APU Fault Panel Installation</td>
<td>FAA STC: ST09308SC-D, Revision 1, dated 20 Sep 1991. Drawing List No.: F90-01024, Revision C, dated 12 Aug 1991.</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/06/2008</td>
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<td>1018320</td>
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<td>EASA:IMA.S.02903</td>
<td>DASSAULT FALCON JET CORP. (SERVICE CENTER)</td>
<td>REV 1</td>
<td>FAA STC SAT225SW-D</td>
<td>MYSTERE FALCON 900</td>
<td>Installation of Emergency Power Supply System.</td>
<td>FAA STC: ST09308SC-D, Revision 3, dated 08 Nov 2000. Drawing List No.: F90-01036, Revision E, dated 03 Nov 2000.</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/06/2008</td>
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<td>EASA:IMA.S.02904</td>
<td>DASSAULT FALCON JET CORP. (SERVICE CENTER)</td>
<td>REV 1</td>
<td>FAA STC SAT227SW-D</td>
<td>MYSTERE FALCON 900</td>
<td>Installation of Air Distribution System Modification.</td>
<td>FAA STC: ST09308SC-D, Revision 4, dated 09 Sep 1991. Drawing List No.: F90-01031, Revision T, dated 26 Jan 1991. AFM Supplement No.: 36098-1, dated 20 Jan 1989.</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA:IMA.S.02906</td>
<td>DASSAULT FALCON JET CORP. (SERVICE CENTER)</td>
<td>REV 1</td>
<td>FAA STC ST09306SC-D</td>
<td>MYSTERE FALCON 900</td>
<td>Installation of Universal Bracket Attachments.</td>
<td>FAA STC: ST09306SC-D, Revision 0, dated 26 Jan 2003. Drawing List No.: F90-01081, Revision A, dated 20 Jan 2000.</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>23/06/2008</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
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<td>FAA STC ST03579AT-D</td>
<td>GULFSTREAM G-V-X</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 4123.</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Master Index List GC141414123 Rev. C, dated July 31, 2008 or later approved revisions.</td>
<td>1. Aircraft Serial Number 4123 only 2. The following STCs are a required part of this STC - ST03027AT-D - ST02987AT-D - ST03047AT-D - ST02968AT-D - ST02959AT-D - ST03032AT-D - ST03039AT-D - ST03032AT-D - ST02979AT-D - Dit is the installer’s responsibility to ensure that these STCs are all EASA3...</td>
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<td>SAVANNAH AIR CENTER, L.L.C.</td>
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<td>FAA STC SA02689AT-D</td>
<td>GULFSTREAM IV</td>
<td>Installation of L3 Communications Digital Flight Data Recorder With Teledyne Digital Flight Data Acquisition Unit.</td>
<td>To be operated in accordance with Approved Flight Manual supplement ref GY/1132-FTM-001, dated July 9, 2003, or later approved revision</td>
<td>Only approved for Aircraft manufactured before 18 August 2000</td>
<td>03/06/2008</td>
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<td>MACHINEING PROGRAMMING MANUFACTURIN Glnc.</td>
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<td>FAA STC SA1033WI</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Replacement of original main landing gear doors with one piece monolithic machined gear doors.</td>
<td>Installation in accordance with FAA approved, MPM Inc. Installation Drawing ref. LG-DR002 revision &quot;norma&quot; or later approved revisions.</td>
<td>1. The attached AML is applicable only on aircraft serial numbers called out in MPM Inc. Drawing No AML-LG-DR002-1, -2, -3 and +fr for installation and applicability. 2 Prior to installation of this modification the installer must determine that the interrelationship between this modification and...</td>
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<td>MACHINEING PROGRAMMING MANUFACTURIN Glnc.</td>
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<td>FAA STC SA00902WI</td>
<td>SEE EASA APPROVED MODEL LIST</td>
<td>Replacement of original main landing gear doors with monolithic machined gear doors.</td>
<td>Installation in accordance with FAA approved, MPM Inc. Installation Drawing ref. LG-DR0021 and Installation Instructions MPAH-103 on aircraft listed in the attached Approved Model List (AML) in MPM Inc, Drawing Number AML-LG-DR002. Rev A dated November 20, 2000, or later approved revision.</td>
<td>1. The attached AML is applicable only on aircraft serial numbers called out in MPM Inc. Drawing No. AML-LG-DR0011. The attached AML is applicable only on aircraft serial numbers called out in MPM Inc Drawing No. AML-LG-DR011.2 Prior to installation of this modification the installer must dete...</td>
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<td>GULFSTREAM AEROSPACE CORPORATION</td>
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<td>FAA STC ST00394LA-D</td>
<td>GULFSTREAM G-V-SF</td>
<td>Installation of Cabin Interior Modification for Aircraft Serial 5155.</td>
<td>Install in accordance with Gulfstream Aerospace Report GC52000K569, Rev. N/C dated 29 July 2008 and Index List GC52000M091, Revision 0 dated 28 July 2008, or later approved revisions.</td>
<td>Aircraft Serial Number 5155 only.</td>
<td>04/06/2008</td>
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<td>DELTA ENGINEERING CORPORATION</td>
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<td>FAA STC ST02402NY-D</td>
<td>AIRBUS A320</td>
<td>Panasonic IFE a eFXInstallation of Panasonic IFE systemEASAInstallation of FAA STC No. ST02402NY-D from 28. March 2008.</td>
<td>FAA STC No. ST02402NY-D from 28. March 2008 Delta Engineering Master Data List 0164-06179-2165 Revision -1, from 15. March 2008, applicable for A320-200 series later revisions other than listed documents approved by EASA in accordance with EASA ED decision 2004/4/C (or subsequent revisions of...)</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.1 The installation of this modification by third pers...</td>
<td>16/07/2008</td>
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<td>JAMCO AMERICA, INC.</td>
<td>FAA STC ST00340SE; P. GILLESPI</td>
<td>BOEING 747-200</td>
<td>Installation of Jamco Gallei in accordance with Jamco America Inc. Master Drawing List LA-036, Rev. A dated September 1, 1995, or later approved revisions.</td>
<td>None.</td>
<td>Prior to installation of this modification, the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The interior re-arrangement, along with required placard...</td>
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<td>EASA/IMA.S.02920</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC SA04-113</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Interior Completion - ACSN 9234</td>
<td>TCCA STC No. SA04-113 Issue 1 dated 28 October 2004 TCCA SOC letter C-07-0590 dated 03 July 2008 Aero Consulting Services Modification Data Summary DS94000 Revision A1 dated 15 November 2007 Bombardier Airplane Flight Manual Supplement G-FM25004001 Revision B Instructions for Continued Airworthiness...</td>
<td>Restricted to ASN 9234 and 1. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOMS.2 WLAN system is compatible only to IEEE 802.11(b).3. Prior to installation of this modification the installer must determine that the internrelationship between this modification...</td>
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<td>EASA/IMA.S.02921</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC SA06-43 ISSUE 2</td>
<td>BOMBARDIER BD-700-1A11 BOMBARDIER BD-700-1A10</td>
<td>Installation of XM Satellite Radio Antenna.</td>
<td>Transport Canada STC No. S06-43-3 ISSUE 2 dated 10 November 2006. Transport Canada Statement of Compliance ref. C-06-0885 dated 21 July 2008 Aero Consulting Services Modification Data Summary no. E821000 Revision C dated 20 October 2006 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person...</td>
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<td>EASA/IMA.S.02924</td>
<td>EAD INC.</td>
<td>FAA STC ST01481CH</td>
<td>BOEING 737-400 SERIES</td>
<td>TCAS Mode-S Activation on Boeing 737-400 Series as approved under the STC FAA ST01481CH.</td>
<td>1. Europe Aero Design Master Data List Doc. No. DC104-0245-00 Rev.2 dated June 26, 2001*2. FAA approved Airplane Flight Manual Supplement, Europe Aero Design Doc. No. DC383-1483-00 Rev.0 approved June 27, 2001** or later revisions off the above listed documents approved by EASA in accordance with Battery Management System and Inertial Reference Sys...</td>
<td>Prior to installation of this modification the installer must determine that the internrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person...</td>
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<td>EASA/IMA.S.02925</td>
<td>CANARD AEROSPACE CORPORATION</td>
<td>FAA STC ST01117CH</td>
<td>BOEING 747-8R BOEING 747-7SP BOEING 747-300 BOEING 747-200F BOEING 747-200C BOEING 747-200 BOEING 747-108B SUD BOEING 747-100</td>
<td>Option 1: Installation of Honeywell Classic Navigator system in accordance with Canard Aerospace Corporation Concept Drawing Nr. 30011, Rev. NC dated November 25, 2002. Appendix A, or later FAA approved revision. Option 2: Installation of a CMA-900 Flight Management System and Inertial Reference Sys...</td>
<td>Option 1: 3.4-3249-04-NC34-3249-00-NCOption 2: 3.4-3275-04-NC 3.4-3275-04-NC 3.4-3275-04-NC 3.4-3275-04-NC 3.4-3275-04-NC 3.4-3275-04-NC 3.4-3275-04-NC 3.4-3275-04-NC 3.4-3275-04-NC 3.4-3275-04-NC</td>
<td>1. Prior to installation of this modification the installer must determine that the internrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person...</td>
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<td>EASA.IM.A.S.02926</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0291LA-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5193.</td>
<td></td>
<td>Install in accordance with Gulfstream Aerospace Report GCS2000K581, Rev. N/C and Index List GCS20008613 Rev. H, dated 20 August 2008, or later approved revisions.</td>
<td></td>
<td>Aircraft Serial Number 5193 only. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of...</td>
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<td>EASA.IM.A.S.02927</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0292LA-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5198.</td>
<td></td>
<td>Install in accordance with Gulfstream Aerospace Report GCS2000K595, Rev. A and Index List GCS20008618 Rev. L, dated 11 September 2008, or later approved revisions.</td>
<td></td>
<td>1. Aircraft Serial Number 5198 only. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation...</td>
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<td>EASA.IM.A.S.02930</td>
<td>HAWKER BEECHCRAFT SERVICES</td>
<td>FAA STC ST0191WI</td>
<td>BEECH 1900D</td>
<td>Installation of Cargo Conversion Modifications.</td>
<td></td>
<td>Installation in accordance with Raytheon Aircraft Services Master Drawing List RAS 0422-1, Revision N, dated July, 1, 2003 or later approved revisions. Operation in accordance with Raytheon Aircraft Services Airplane Flight Manual Supplement Number 05-422, dated August 26, 2003 or later approved revisions.</td>
<td></td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously approved configuration will not introduce any adverse effect upon the airworthiness of the product. The approval of this modification applies to the...</td>
<td>03/09/2008</td>
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<td>EASA.IM.A.S.02931</td>
<td>HAWKER BEECHCRAFT SERVICES</td>
<td>REV. 1 FAA STC ST0586LA</td>
<td>HAWKER 1000</td>
<td>Installation of AlliedSignal Enhanced Ground Proximity Warning System (EGPWS).</td>
<td></td>
<td>Raytheon Aircraft Services Master Data List for Allied Signal Enhanced Ground Proximity Warning System Installation on Hawker 1050A Aircraft Report No. RAY-R3-002, Revision &quot;A&quot; dated April 3, 1998. To determine which configuration(s) or the EGWS have been approved, refer to the appropriate sections of...</td>
<td></td>
<td>The installation should not be incorporated in any aircaft unless it is determined that the relationship between this installation and any previously approved configuration will not introduce any adverse effect upon the Airworthiness of the Aircraft. The approval of this modification applies to the...</td>
<td>04/07/2008</td>
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<td>EASA.IM.A.S.02932</td>
<td>SAFFRAN CABIN GALLEYS US, INC.</td>
<td>BOEING 747-200C BOEING 747-200B</td>
<td>Installation of a TBO Container.</td>
<td>Driessen Aircraft Interior Systems Data List MDL-101996, Revision &quot;C&quot; dated April 14, 1998.</td>
<td></td>
<td>The installation should not be incorporated in any aircaft unless it is determined that the relationship between this installation and any previously approved configuration will not introduce any adverse effect upon the Airworthiness of the Aircraft. The approval of this modification applies to the...</td>
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<td>ARMSTRONG AEROSPACE Inc.dba ASTRONICS ARMSTRONG AEROSPACE</td>
<td>FAA ST02536CH</td>
<td>A330-341, A330-342, A330-343, A330-321, A330-322, A330-323, A330-301, A330-302, A330-303</td>
<td>Activation of Honeywell Runway Awareness and Advisory System (RAAS).</td>
<td>FAA STC ST02536CH, issued 17th March 2008.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>03/09/2008</td>
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<td>ARMSTRONG AEROSPACE Inc.dba ASTRONICS ARMSTRONG AEROSPACE</td>
<td>FAA ST02537CH</td>
<td>A340-642, A340-311, A340-312, A340-313</td>
<td>Activation of Honeywell Runway Awareness and Advisory System (RAAS).</td>
<td>FAA STC ST02537CH, issued 17th March 2008.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>GULFSTREAM AEROSPACE CORPORATION.</td>
<td>FAA STC ST10338SC-D</td>
<td>GULFSTREAM G150</td>
<td>Installation of an Emergency Power/Annunciator System i.a.w. Gulfstream Index List GA313318000, Rev. B or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>ICA GA34331AG00AFMS GA31331M00</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>GULFSTREAM AEROSPACE CORPORATION.</td>
<td>FAA STC ST10338SC-D</td>
<td>GULFSTREAM G200</td>
<td>Installation of an XM Satellite Radio System i.a.w. Index List GA216078000, Rev. A. or later approved revisions of the aforementioned document approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>ICA GA 24607A000</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.A.S.02942</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>REV 1</td>
<td>FAA STC SA00345E</td>
<td>CESSNA T182</td>
<td>Installation of vortex generators on the wings and tail surfaces in accordance with the provisions of AML SA00345E.</td>
<td>Approval of this change in type design applies to the aircraft listed on AML SA00345E only. This approval should not be extended to other aircraft on which other previously approved modification are incorporated unless it is determined that the relationship between this change and any other previou...</td>
<td></td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification b...</td>
<td>27/10/2008</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST0357AT-D</td>
<td>GULFSTREAM GV-X</td>
<td>Installation of an Airplane Interior for Aircraft Serial 4130.</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC414914130 Rev E, dated October 1, 2008, or later approved revisions. Aircraft Serial Number 4130 only. The following STCs are a required part of this STC:ST03027AT-DST02987AT-DST03047AT-DST02980AT-DST03290AT-DST03532AT-DST02979AT-DST03979AT-Gt...</td>
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<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03592AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5201.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List, GC514915201 Rev G, dated October 22, 2008, or later approved revision. To be operated in accordance with FAA approved Airplane Flight Manual Supplement, refGC51501M029, Rev NC, dated 22 October 2008 or later approved. Prior to installation of this modification the installer must...</td>
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### List of EASA Supplemental Type Certificates

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<td>EASA.IM.A.S.02954</td>
<td>CASCADE AEROSPACE Inc.</td>
<td>TCCA STC- NUMBER SA08-91</td>
<td>BOMBARDIER DC-8-400 SERIES</td>
<td>DHC-8-400 Conversion to Bulk Freighter</td>
<td>TCCA STC Number SA08-91, issue 1, dated 31 December 2008</td>
<td>Installation of the freighter conversion must be in accordance with Cascade Aerospace Inc. Eq 4079, Rev. 2, TCCA approved 12 December 2008 and Master Document List MDL 4079, Rev. 2, TCCA approved 24 December 2008.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>20/12/2009</td>
<td>Active</td>
<td>1011652</td>
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<td>EASA.IM.A.S.02955</td>
<td>B/E AEROSPACE, Inc. - FSI</td>
<td>FAA ST00365E</td>
<td>747-200</td>
<td>This design change relocates the E/E Bay Access Hatch on the main deck floor and installs Garci o Handling System Provisions for a 30 pallet main deck cargo configuration per Flight Structures, Inc.</td>
<td>Master Drawing list 96FS24-001 at Revision E dated September 18, 2003, or later FAA approved revisions.</td>
<td>This STC can only be installed in aircraft with FAA STC SA2725CE-D or STC ST00255WI-D or STC ST00185WI-D or STC ST001845SE installed.</td>
<td>18/09/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.02957</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC C-LSA03-372/D</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Aircraft Completion, 13 Passenger Interior Configuration</td>
<td>Transport Canada Soc ref. C-08-0754 dated 13 November 2008. Transport Canada STC No. C-LSA03-372/D Issue 2 dated 13 November 2008. Aero Consulting Services Modification Data Summary D58300 Revision A dated 13 November 2008 or later approved revision. Required Flight Manual Supplements The following app...</td>
<td>Restricted to ASN 9120. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification...</td>
<td>15/12/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.02958</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC C-LSA01-417/D</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Approval of Side Facing Divan Limitations.</td>
<td>Transport Canada STC No. C-LSA01-417/D Issue 2 dated 16 January 2004. TCCA Soc reference C-04-0404 dated 16 September 2008. Aero Consulting Services Modification Data Summary No. D14000 Revision A or later approved revision. Aero Consulting Services Flight Manual Supplement D14091 Rev A or later appro...</td>
<td>Restricted to ASN 9016. Required Equipment Previous installation of corresponding EASA STC No. EASA.IM.A.S.01252 (FAA STC SA8164NM-D) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>19/06/2008</td>
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<td>EASA.IM.A.S.02959</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST0307AT-D</td>
<td>GV</td>
<td>Installation of Honeywell Integrated Avionics Computers (IAC) with Software Version NZ 6.0.</td>
<td>Installation of Honeywell Integrated Avionics Computers (IAC) with Software Version NZ 6.0 for -501 and -503 configurations in accordance with Gulfstream Aerospace Corporation Index List CES1 118801 Rev. D or later approved. Operation in accordance with Gulfstream Airplane Flight Manual Supplement C...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/10/2008</td>
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<td>Description</td>
<td>Certification basis</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.02960</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03074AT-D</td>
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<td>GV</td>
<td>Installation of Honeywell NZ-2000 Third FMS Computer with Software Version NZ 6.0.</td>
<td>Installation of Honeywell NZ-2000 Third FMS Computer with Software Version NZ 6.0 in accordance with Gulfstream Aerospace Corporation Index CE51 3088000 Rev C or later approved. Operation to be in accordance with Gulfstream Airplane Flight Manual Supplement CE51 3088000 Rev. NC Dated April 15, 20...</td>
<td>Prior to installation of this modification the installer must determine that the modification will introduce no adverse effect upon the airworthiness of the product. Prior to installation of this modification the installer must determine that the modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/10/2008</td>
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<td>10016569</td>
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<td>EASA.IM.A.S.02967</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST00949WK-D</td>
<td>LEARJET 60</td>
<td>Cockpit Speaker Mute Function FAA STC ST00949WK-D</td>
<td>As per Master Document List M100339ICT, rev. G (or later approved revisions)</td>
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<td>06/10/2008</td>
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<td>EASA.IM.A.S.02970</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCIA STC C-LSA03-334D</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Installation of interior furnishings and cabin systems, applicable to BD-700-1A10 serial number 9112 only.</td>
<td>TCIA STC C No. C-LSA03-334D Issue 2 dated 06 October 2003 TCIA SOC Reference C-08-0755 dated 16 September 2008 Aero Consulting Services Modification Data Summary D42500 Revision B or later approved revision. The following Flight Manual Supplements are required G-FM24000039 Rev N/C...</td>
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<td>EASA.IM.A.S.02971</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC C-LSA05-100/D</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Relocation of AB Toilet Compressor.</td>
<td>Transport Canada STC No. C-LSA05-100/D Issue 1 dated 17 March 2005.TCCA SOC reference C-08-0755 dated 16 September 2008.Aero Consulting Services Modification Data Summary No. E396500 Revision NC dated 17 March 2005 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>19/02/2008</td>
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<td>EASA.IM.A.S.02972</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01421W-0</td>
<td>LEARJET 60</td>
<td>Arrow 410 System</td>
<td>As per Master Data List M100476ICT, rev. P. (or later approved revisions).</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>16/09/2008</td>
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<td>EASA.IM.A.S.02973</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01424W-0</td>
<td>LEARJET 60</td>
<td>Emergency power supply 1</td>
<td>As per master Data List M100487ICT, rev. M (or later approved revisions).</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>16/09/2008</td>
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<td>EASA.IM.A.S.02975</td>
<td>HAWKER BEECHCRAFT CORPORATION (HBC)</td>
<td>FAA STC ST01113LA</td>
<td>BEECH 400A</td>
<td>Rockwell Collins dual Altitude Heading Reference system (AHRS).</td>
<td>None.</td>
<td>Cutter aviation Master Data List Document No. 02A-2408-201 Revision &quot;B&quot; dated December 5, 2007 or later approved revisions.</td>
<td>12/09/2008</td>
<td>Active</td>
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<td>EASA.IM.A.S.02985</td>
<td>RAIBBECK ENGINEERING Inc.</td>
<td>FAA STC SA3031NM</td>
<td>BEECH B200T BEECH B200CT BEECH B200C BEECH B200 A200CT B200 A200 B200CT B200C B200 200T 200CT 200C 200</td>
<td>Installation of improved leading edge consisting of inboard leading edges, deice boots, leading edge ducts, leading edge to nacellefairing, and associated hardware.</td>
<td>1. Installed in accordance with Raabeck installation manual Installation Manual R81-1032, Rev. F, dated January 30th 2008.2. Instructions for Continued Airworthiness Ref. Raabeck Maintenance Manual 85-120 Rev. G, dated January 2008.3. IPC 85-119 dated April 2004.4. Or later approved revision...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>05/08/2008</td>
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<td>EASA.IM.A.S.02988</td>
<td>RAIBBECK ENGINEERING Inc.</td>
<td>FAA STC SA3369NM</td>
<td>B200T B200CT B200CT B200CT B200CT B200C B200 A200CT A200CT A200 200T 200CT 200C 200</td>
<td>Installation of Ram Air Recovery System.</td>
<td>1. Installed in accordance with Raabeck installation manual Installation Manual R81-1031 Rev. 1, dated February 8, 2006.2. Approved Flight Manual Supplement 99-117 Revision, dated February 1, 2008.3. Instruction for Continued Airworthiness ref. Raabec Maintenance Manual 85-120 Rev. G.4...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>05/08/2008</td>
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<td>EASA.IM.A.S.02989</td>
<td>GULFSTREAM AEROSPACE CORPORATION.</td>
<td>FAA STC ST010356C-D</td>
<td>GULFSTREAM 200</td>
<td>Installation of Executive Cabin Interior i.a.w. Gulfstream Index List GA200008053, Rev. A or later revisions of this document approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>ICA GA210025003FM/G GA214688F063</td>
<td>Limited to S/N 063</td>
<td>20/02/2008</td>
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<td>EASA.IM.A.S.02990</td>
<td>DASSAULT FALCON JET CORP.(SERVICE CENTER)</td>
<td>FAA STC SA8271SW-D</td>
<td>FALCON 900EX</td>
<td>EASA Validation of FAA STC SA82715W-D: Installation of the APU Fault Panel Installation i.a.w. Drawing List Report No. F9EX-0107 dated 08th October 1996 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions).</td>
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<td>EASA.IM.A.S.02991</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC ST093135C-D</td>
<td>FALCON 900EX</td>
<td>EASA Validation of FAA STC ST093135C-D Revision 1, 01st October 2002 Installation of Standby Horizon System Modifications in accordance with Drawing List Report No. F9EX-01035 Revision B dated 25 September 2002 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions).</td>
<td>Free of Charge</td>
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<td>EASA.IM.A.S.02992</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC ST093115C-D</td>
<td>FALCON 900EX</td>
<td>EASA Validation of FAA STC ST093115C-D Revision 1, 01st October 2002 Installation of the Inertial Reference System Modification in accordance with Drawing List Report No. F9EX-01033 Revision NC dated 04th May 2000 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions).</td>
<td>See those of original FAA STC SA8291SW-D: Installation of the Engine Instrument Display System Modification in accordance with Drawing List Report No. F9EX-1034 Revision NC dated 04th May 2000 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions).</td>
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<td>EASA.IM.A.S.02994</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC SA8201SW-D</td>
<td>FALCON 900EX</td>
<td>EASA Validation of FAA STC SA8201SW-D: Installation of the Battery Installations i.a.w. Drawing List Report No. F9EX-01003 Revision F dated 14 September 2000 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions).</td>
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<td>EASA.IM.A.S.02995</td>
<td>DASSAULT FALCON JET CORP.(SERVICE CENTER)</td>
<td>FAA STC ST093125C-D</td>
<td>FALCON 900EX</td>
<td>EASA Validation of FAA STC ST093125C-D: Installation of the Engine Instrument Display System Modification in accordance with Drawing List Report No. F9EX-1034 Revision NC dated 04th May 2000 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions).</td>
<td>See those of original FAA STC ST093135C-D: Installation of the Engine Instrument Display System Modification in accordance with Drawing List Report No. F9EX-1034 Revision NC dated 04th May 2000 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions).</td>
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<td>EASA.IM.A.S.02998</td>
<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>FAA STC SA8275SW-D</td>
<td>FALCON 900EX</td>
<td>EASA Validation of FAA STC ST093125C-D: Installation of the Engine Instrument Display System Modification in accordance with Drawing List Report No. F9EX-1034 Revision NC dated 04th May 2000 or later revisions approved by EASA i.a.w. EASA ED Decisions 2007/001/C and 2004/04/CF (or subsequent revisions to these decisions).</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.A.S.03001</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
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<td>TCCA STC SA04-113</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Customized Interior Completion - ACSN 9270.</td>
<td>TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. TCCA SOC reference C-08-0433 dated 17 November 2008. Aero Consulting Services Configuration Deviation List D054200 Revision BX dated 14 November 2008. Instructions for Continued Airworthiness GC92700000-IFCA.</td>
<td>Restricted to ASN 9270 and T. Refer to the Limitations sections of the required AFM Supplement and to the referenced FCOM/MS.2.WLAN system is compatible only to IEEE 802.11(b). Prior to installation of this modification the installer must determine that the interrelationship between this modification...</td>
<td>21/11/2008</td>
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<td>EASA.IM.A.S.03003</td>
<td>FLINT AERO, INC.</td>
<td></td>
<td>CESSNA TR182 CESSNA T182, T182T CESSNA R182</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Auxiliary Internal Wing Fuel Tanks This STC is a revision of LBA STC SA 1305, dated 14 January 2003.</td>
<td>None.</td>
<td>None.</td>
<td>15/08/2008</td>
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<td>EASA.IM.A.S.03004</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>REV. 1</td>
<td>TCCA STC SA07-019</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Waterproofing and Spill Management, Interior Modifications</td>
<td>TCCA STC No. SA07-19 Issue 3 dated 25 July 2008. TCCA SOC reference NA05012-07-0177 dated 27 February 2009. Aeromix Aviation Inc. Master Document List - Waterproofing and Spill Management, Interior Modifications - Bombardier BD-700 Aircraft. Document No. MDL26141002, Revision F, dated 27 January 2009. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>16/03/2009</td>
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<td>EASA.IM.A.S.03005</td>
<td>U.S. TECHNICAL CONSULTANTS, INC.</td>
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<td>FAA STC ST02409CH-D</td>
<td>DC-9-87 (MD-87) DC-9-83 (MD-83)</td>
<td>FAA STC No. ST02409CH-D &quot;NavAero i-BagC22 Electronic FlightBag&quot;</td>
<td>MEL 469A-23-009 Rev. F dated 13 August 2007. PM Supplement U10-0163 dated 6 November 2007 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>16/01/2009</td>
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<td>EASA.IM.A.S.03006</td>
<td>MITSUBISHI HEAVY INDUSTRIES AMERICAN, INC.</td>
<td></td>
<td>FAA STC SA17515W</td>
<td>MU-2B, -10, -15, -20, -25, -26A, -35, -36, -36A, -40</td>
<td>Flight Crew Restraint System.</td>
<td>FAA STC SA17515W Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>22/09/2008</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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## List of EASA Supplemental Type Certificates

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<td>EASA.IM.A.S.03010</td>
<td>S-TEC CORPORATION</td>
<td>FAA STC: SA0257AC-D</td>
<td>PIPER PA-28-210, PIPER PA-28-180, PIPER PA-28-160, PIPER PA-28-150, PIPER PA-28-140</td>
<td>Installation of S-TEC System 30 ALT Attitude Hold System, Model ST-712 (14 Volt System)</td>
<td>FAA approved S-TEC Bulletin No. 812, dated 06-July-1997, Initial Issue, or later approved revision.</td>
<td>Prior to installation of this modification the installer must verify the availability of EASA approval for one or more mentioned STC(s) when the product is modified per that/these STC(s).1 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>14/11/2009</td>
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<td>EASA.IM.A.S.03022</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA SA00-109, ISSUE 4</td>
<td>DHC-8-200, DHC-8-102, DHC-8-103</td>
<td>Installation of Environmental Control System, Low Temperature Limit System (Cold Clamp) Ref. TCCA STC SA00-109, Issue 4, dated 19 October 2000</td>
<td>Modification has to be completed in accordance with ACS Modification Data Summary, C-46400, Rev. G, dated 05-July-2009 ACS Maintenance Manual Supplement, C46481, Rev. A, dated 15 January 2007 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.A.S.03023</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA SA02-4, ISSUE 1</td>
<td>CL-600-2B19</td>
<td>Installation of NAI part 093602 Solid State Logo Light</td>
<td>Transport Canada STC No. SA02-4 Issue 1 dated 9 January 2002. Transport Canada STC Summary GC20008128. Revision B or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>14/10/2008</td>
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<td>EASA.IM.A.S.03024</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC: ST03817AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of Export Modifications for Aircraft Serial 4091.</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Export Modifications Index List, GC420008128, Rev C, dated October 7, 2008, or later approved revisions.</td>
<td>Aircraft Serial Number 4091 only.2 This STC must be installed in conjunction with Aircraft Completion STC ST0347AT-G (EASA.IM.A.S.03025) to fully meet EASA requirements.3 It is the installers' responsibility to ensure that all STC's relevant to fully meet EASA approval 4. Prior to installation of this modification the installer must...</td>
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<td>EASA.IM.A.S.03025</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC: ST0347AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Airplane Completion and Custom Cabin Installation for Aircraft Serial 4091.</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Master Index List, GC414914091, Rev C, dated October 30, 2007, or later approved revisions.</td>
<td>Aircraft Serial Number 4091 only. The following STCs are a required part of this STC: ST03027AT-D, ST03028AT-D, ST03029AT-D, ST03030AT-D, ST03031AT-D, ST03032AT-D, ST03033AT-D, ST03034AT-D, ST03035AT-D.</td>
<td>Prior to installation of this modification the installer must...</td>
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<td>EASA.IM.A.S.03026</td>
<td>INFLIGHT CANADA Inc.</td>
<td>TCCA STC: CL-00-127/D</td>
<td>BOMBARDIER CL-600-2B19</td>
<td>Honeywell SCS-1000 Mini-M Aero SATCOM System and Securaplane Camera Installation.</td>
<td>Transport Canada Stc No. 5010-O-08-0561 dated 22 October 2008.</td>
<td>1. DECA Aviation / Inflight Canada Modification Summary MS03555, Revision 4, dated 16 October 2004, or later approved revision.2 DECA Aviation / Inflight Canada Instructions for Continued Airworthiness ISA03555, Revision 2, dated 12 October 2004, or later approved revision.3 Prior to installation...</td>
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<td>LEARJET Inc.</td>
<td>FAA STC ST01370W-D</td>
<td>LEARJET 80</td>
<td>Honeywell HF Communication System.</td>
<td>As per Master Document List M100497CT, rev. V or later revisions approved by EASA in accordance with EASA ED decision 2000/04/04 (or subsequent revisions of this decision).</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA.IM.A.S.03032</td>
<td>THUNDER AVIATION SERVICES INC.</td>
<td>FAA STC ST01695LA</td>
<td>MYSTERE-FALCON 20-ES, FS, MYSERE-FALCON 20-CS, 2DS, FAN JET FALCON SERIES C, D, E, FAN JET FALCON</td>
<td>EASA Validation of FAA STC STS1695LA Installation of Rosemount Pitot-Static Probes, Part Numbers 0856L01 and 0856L02 and Honeywell Air Data System comprised of dual AZ-252 Air Data Computers, dual BA-250 Attitude Displays and AL-800 Attitude Alerting Unit, w. Thunder Aviation Inc Master Drawing List...</td>
<td>FAA STC ST01695LA, as per FAA amendment 04th September 2007.</td>
<td>18/05/2009</td>
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<td>EASA.IM.A.S.03033</td>
<td>BVE AEROSPACE LIMITED</td>
<td>TCCA C-LSA08-268D</td>
<td>BD-700-1A10</td>
<td>Installation, Side Facing Divan, S/N 9260 Configuration</td>
<td>TCCA STC No. C-LSA08-268D Issue 1 dated 18 September 2008. TCCA SC of ref. C-08-0759 dated 28 November 2008. Aero Consulting Services Ltd. Modification Data Summary Number F957000 Revision NEC or later approved revision. Aero Consulting Services Ltd. Airplane Flight Manual Supplement F957000 Revision N-</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>13/03/2009</td>
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<td>EASA.IM.A.S.03038</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST03943AT-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Aircraft completion and installation of Airplane Interior for Aircraft Serial 5210.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Index List GC514915210, Rev. G, dated December 10, 2008 or later FAA approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.A.S.03039</td>
<td>ROEMOUNT AEROSPACE, Inc.</td>
<td>FAA STC ST01474CH</td>
<td>MD-11, MD-11F</td>
<td>Installation of Rosemount Aerospace Model 1535D Pressure Transducer Inconformance with Rosemount Aerospace Master Data List, Document no. 20100401, Revision C, dated June 1, 2001 or later FAA approved revision.</td>
<td>None.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.A.S.03041</td>
<td>UNIVERSAL AVIONICS SYSTEMS CORPORA TION</td>
<td>FAA STC ST01303LA</td>
<td>BOMBARDIER CL-600-2B16 BOMBARDIER CL-600-2A12 (CL-601 BOMBARDIER CL-600-1A11 (CL-600</td>
<td>Installation of Universal Avionics UNS-1F/ L and optional Unilink communication system</td>
<td>FAA STC No. ST01303LA dated 25 August 2008. Universal Avionics Systems Master Drawing List, Document No. MDA-954, Revision G, dated 25 September 2007 or later approved revision.</td>
<td>1. Restricted to the following configurations: -3 Config: UL-711 with SCN 22.X Config: UNS-1F with SCN 802.X and AT02 Maps2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/12/2009</td>
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<td>EASA.IM.A.S.03042</td>
<td>UNIVERSAL AVIONICS SYSTEMS CORPORA TION</td>
<td>FAA STC ST01141LA</td>
<td>BOMBARDIER CL-600-2B16 BOMBARDIER CL-600-2A12 (CL-601 BOMBARDIER CL-600-1A11 (CL-600</td>
<td>Installation of Universal Avionics Terrain Awareness Warning System (TAWS)</td>
<td>FAA STC No. ST01141LA dated 22 February 2008. Universal Avionics Master Drawing List, MDA-9068, Revision G, dated 30 November 2008, or later approved revision.</td>
<td>1. Restricted to the following configuration: -4 Config: Terrain Awareness and Warning System (TAWS) operating software SCN 11.X (ETSO EASA.IM.210.0170) displayed on MFD-640 (ETSO EASA.IM.210.0170) displayed on MFD-640. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>07/01/2009</td>
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<td>EASA.IM.A.S.03049</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01402WI-D</td>
<td>LEARJET 60</td>
<td>Collins Pro Line 21 Avionics Installation FAA STC ST01402WI-D</td>
<td>As per Master Data List AEC-0497-001, rev. L (or later approved revisions)</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA.IM.A.S.03050</td>
<td>LEARJET Inc.</td>
<td>FAA STC ST01422WI-D</td>
<td>LEARJET 60</td>
<td>Interior Upgrade (Floor Plans A, B, C &amp; D)</td>
<td>FAA STC ST01422WI-D</td>
<td>As per Master Document List M100471ICT, rev. BH (or later approved revisions)</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/11/2008</td>
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<td>EASA.IM.A.S.03051</td>
<td>RAISBECK ENGINEERING Inc.</td>
<td>FAA STC SA00752SE</td>
<td>SEE ATTACHED EASA APPROVED TYP BEECH 200</td>
<td>This change applies only to the above listed model aircraft in accordance with Raisbeck Engineering Master Drawing List 05-1500, Revision D, dated May 13, 2008 or later approved revisions.</td>
<td>Operation in accordance with FAA Approved Flight Manual Supplement 98-126, Revision A, dated January 10, 2000, 98-126H, dated October 4, 2007, or 08-BD007-G-126, Revision I, dated May 13, 2008, or later approved revisions.</td>
<td>Requires the prior incorporation of Raisbeck STC's ref: SAA11754M (equipped with High Floatation gear) SA23695NAS3511NAS3519 RM1 is the installer's responsibility to ensure all these referenced STC's are EASA approved.</td>
<td>15/12/2008</td>
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<td>EASA.IM.A.S.03052</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST0032CH-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Aircraft completion and cabin interior installation for Serial No 5212.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC52 2046Y121, Revision B, dated February 5, 2009, or later approved revisions. Operation to be in accordance with Gulfstream Flight Manual Supplement GC52 2048X038, Original Release, dated February 15, 2009, or later.</td>
<td>1. Serial No. 5212 Only. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>02/03/2009</td>
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<td>EASA.IM.A.S.03053</td>
<td>LITEAIR AVIATION PRODUCTS Inc.</td>
<td>FAA STC ST01872SE</td>
<td>BOMBARDIER DHC-8-402 BOMBARDIER DHC-8-401 BOMBARDIER DHC-8-400 SERIES</td>
<td>Installation of lightweight window plug. Ref. FAA STC ST01872SE, dated 09 September 2008.</td>
<td>1. Installation in accordance with LiteAir Aviation Products Inc. Master Drawing List LAP1006-D01, Original Issue, dated 01 August 2008. 1.2. Maintenance must be performed in accordance with Light Weight Window Plug Installation. Document LAP0208, Original Issue, dated 07 August 2008.</td>
<td>03/12/2008</td>
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<td>EASA.IM.A.S.03054</td>
<td>AVIATION CONSULTING &amp; ENGINEERING SOLUTIONS, INC.</td>
<td>FAA STC SA01282WI</td>
<td>525</td>
<td>Installation of Replacement of Seat Cushions.</td>
<td>- Cushions must be installed iaw. to Aviation Consulting and Engineering Solutions Master Data List Doc. No. A0819404 - stamped &quot;FAA Approved&quot; from 21 Oct 2004, or later. FAA approved Revisions. Installation only approved on seat assemblies as listed in FAA STC SA01282WI.</td>
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<td>EASA.IM.A.S.03064</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA C-LSA09-087/D</td>
<td>GIV</td>
<td>Retrofit and installation of Cabin Interior Modifications.</td>
<td>Installation in accordance with Aero Consulting Services Ltd Modification Data Summary G004000 Rev NC or later approved revisions. Maintenance in accordance with Aero Consulting Services Ltd Instructions for Continued Airworthiness G004000 Rev NC or later approved revisions.</td>
<td>08/04/2009</td>
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<td>B/E AEROSPACE LIMITED</td>
<td>TCCA C-LSA09-086/D</td>
<td>GIV</td>
<td>Installation of Rockwell Collins Radio Tuning Unit RTU-4210 and Rockwell Collins TDR-40D (ELS/EHS) Transponder.</td>
<td>Aircraft Serial Number 1202 Only Prior to installation of this modification, the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by the operator is responsible for the inclusion of a 24 month check in their maintenance schedule of the Mode S transponder system to ensure correct operation of the 24BIT address and the required DAPs IAW ACS MM 34 53 00.</td>
<td>08/04/2009</td>
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<tr>
<td>BOMBARDIER AEROSPACE</td>
<td></td>
<td>BD-100-1A10</td>
<td>Customized Interior Completion - Aircraft 200659 passenger seats interior configuration. EASA Validation.</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>06/11/2008</td>
<td>Active</td>
<td>1011632</td>
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<td>FAA STC ST10251SC-D</td>
<td></td>
<td>GULFSTREAM 200</td>
<td>Installation of Satcom High Speed Data Antenna i.e. Index List GA243036B000 Rev. D or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04C-FCP (or subsequent revisions of this decision).</td>
<td>1 Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>21/01/2009</td>
<td>Active</td>
<td>1011632</td>
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<tr>
<td>FAA STC ST10355C-O</td>
<td></td>
<td>GULFSTREAM 200</td>
<td>Installation of Thrane &amp; Thrane Aero HSD + Satcom System i.e. Index List GA213038B000 Rev. D or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04C-FCP (or subsequent revisions of this decision).</td>
<td>1 Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>22/01/2009</td>
<td>Active</td>
<td>1011634</td>
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<td>FAA STC ST10251SC-D</td>
<td></td>
<td>GULFSTREAM 200</td>
<td>Installation of THRANE &amp; THRANE AEROSPACE CORPORATION.</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>22/01/2009</td>
<td>Active</td>
<td>1011634</td>
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<td>TCCA STC SA4-113 ISSUE 1</td>
<td></td>
<td>BOMBARDIER AEROSPACE</td>
<td>Installation. Complete Custom Aircraft Interior / Installation of cabin systems and custom furnishings - ACSN 9291</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>20/12/2009</td>
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<tr>
<td>FAA STC ST01635 LA</td>
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<td>AEROCOM ENGINEERING COMPANY</td>
<td>- Installation of Cabin Interior Noise Reduction Kit. Validated by FAA ST1635LA.</td>
<td>1. Restricted to ASN 9291.2. Refer to the Limitations sections of the required AFM Supplement and to the referenced FC/MS.3. WLAN system is compatible only to IEEE 802.11(b).4. Prior to installation of this modification the installer must determine that the relationship between this...</td>
<td>20/01/2009</td>
<td>Active</td>
<td>1011636</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
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<td>EASA.IM.A.S.03077</td>
<td>CESSNA AIRCRAFT COMPANY</td>
<td>FAA STC SA0159WII</td>
<td>CESSNA 525</td>
<td>Installation of an observer seat/3rd audio station/Extended Range Comm.</td>
<td>1. According to Cessna Master Document List Doc. No. CF90004 Rev. B, or later Revisions.2. To be operated in accordance with FAA approved Cessna AFM-S Document No. FM90940, Original Issue, or later approved Revisions.3. To be maintained in addition to basic aircraft maintenance manuals i.a.</td>
<td></td>
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<td>EASA.IM.A.S.03078</td>
<td>AMTECH AERONAUTICAL LIMITED</td>
<td>TCCA STC C-LSA06-378/D</td>
<td>LEARJET MODEL 45</td>
<td>AirCell Anxess II Iridium Satcom TCCA LSTC C-LSA06-378/D</td>
<td>As per Master Document List BAS45010021, rev. E (or later approved revisions)</td>
<td>1. Prior to installation of this modification the installer must determine that the interaction relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by FAA STC ST09293SC-D</td>
<td>06/01/2009</td>
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<td>EASA.IM.A.S.03079</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA FAA STC ST02636CH-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Aircraft Serial No 5220 completion and cabin interior installation.</td>
<td>Installation in accordance with Gulfstream Aerospace Corporation Master Drawing List No. GC32 2048129. Revision A, dated March 26, 2009, or later approved revisions.Operation to be in accordance with Gulfstream Flight Manual Supplement GC32 204M040, Original Release, dated March 26, 2009, or later...</td>
<td>Prior to installation of this modification the installer must determine that the interaction relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>08/04/2009</td>
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<tr>
<td>EASA.IM.A.S.03081</td>
<td>DASSAULT AIRCRAFT SERVICES CORP.</td>
<td>FAA STC ST09293SC-D</td>
<td>FALCON 900EX</td>
<td>EASA Validation of FAA STC ST09293SC-D Installation of a Cabin Master Switch System i.a.w. Drawing List Report No. F9EX-1041</td>
<td>none</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>EASA/IMA.S.0382</td>
<td>DASSAULT FALCON JET CORP. (SERVICE CENTER)</td>
<td>FAA STC ST09015C-D</td>
<td>FALCON 900EX</td>
<td>EASA Validation of FAA STC ST09015C-D Installation of the Service and Indirect Lightning System in accordance with Drawing List Report No. F9EX-01027, Revision B dated 28th September 2000 or later revisions approved by EASA (i.w. EASA ED Decisions 2007/001C and 2004/043/C) (or subsequent revisions to these decisions).</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by...</td>
<td>31/03/2009</td>
<td>Active</td>
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<tr>
<td>EASA/IMA.S.0383</td>
<td>DASSAULT FALCON JET CORP. (SERVICE CENTER)</td>
<td>FAA STC ST09715C-D</td>
<td>FALCON 900EX</td>
<td>EASA Validation of FAA STC ST09715C-D Installation of the HF Radio Master System in accordance with Drawing List Report No. F9EX-01044 Revision NC dated 18 September 2002 or later revisions approved by EASA (i.w. EASA ED Decisions 2007/001C and 2004/043/C) (or subsequent revisions to these decisions).</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by...</td>
<td>31/03/2009</td>
<td>Active</td>
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<td>EASA/IMA.S.0384</td>
<td>DASSAULT FALCON JET CORPORATION</td>
<td>FAA STC ST09715C-D</td>
<td>FALCON 900EX</td>
<td>- EASA Validation of FAA STC ST09715C-D Installation of the Control Wheel Modifications in accordance with Drawing List Report No. F9EX-01032 - FAA STC ST09715C-D Revision 1, 03rd September 2002*. Drawing List Report No. F9EX-01022 Revision C dated 01st August 2002** EASA Approved Airplane Flight Manual Supplement; Report No. 57036-1, dated 01st October 2002 for the F9EX-01044-501 installation or later revisions of these documents approved by EASA (i.w. EASA ED Decisions 2007/001C...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by...</td>
<td>11/05/2009</td>
<td>Active</td>
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<tr>
<td>EASA/IMA.S.0385</td>
<td>THE BOEING COMPANY - WICHITA DIVISION</td>
<td>REV. 1 FAA STC SA124CE-D</td>
<td>BOEING 747-200B</td>
<td>Conversion of Passenger Airplane to Main Deck Side Cargo Door Dedicated Freighter in accordance with Boeing Drawing List document D3-8857, Section V, Appendix 33, Rev. C8. Boeing Commercial Airplane Group - Wichita Division, Airplane Flight Manual Supplement dated 9 September 1992 or later revision approved in accordance with 02004/043/C dated 19 December 2004.D6-30680-7 Cargo Loading Manual</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by...</td>
<td>09/02/2009</td>
<td>Active</td>
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<tr>
<td>EASA/IMA.S.0388</td>
<td>AIRCRAFT TECHNICAL DEVELOPMENT Inc.d/b/a AEROCON ENGINEERING COMPANY</td>
<td>FAA STC ST01568A</td>
<td>737-700</td>
<td>- Installation of Cabin Interior Noise Reduction Kit - Validation of FAA ST01568A. Aerccon Engineering Company Master Document List No. AC-BBJ-1000 Rev. B dated November 18, 2003 or later revisions of the above listed documents approved by EASA in accordance with EASA ED decision 2004/043/C (or subsequent revisions of this decision).</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by...</td>
<td>20/01/2009</td>
<td>Active</td>
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<tr>
<td>EASA/IMA.S.0391</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA C-LSA9-088D</td>
<td>GIV</td>
<td>Installation in accordance with Aero Consulting Services Ltd Modification Data Summary GS02006 Rev NC or later approved revisions. Maintenance in accordance with Aero Consulting Services Ltd Instructions for Continued Airworthiness GS0207 Rev NC or later approved revisions. Aircraft Serial Number 1302 Only/Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>08/04/2009</td>
<td>Active</td>
<td>1001647</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
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<td>ST03595</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST00542AT-D</td>
<td>GULFSTREAM GV-X</td>
<td>Installation of an Aft Galley/Galley Annex Multiple STC.</td>
<td>Install in accordance with Gulfstream Aerospace CorporationIndex List GC41548800 Rev L dated February 04, 2009, or later FAA approved revision.</td>
<td>1. This Aft Galley/Galley Annex Multiple STC, (FAA ref ST00542AT-D), does not approve the entire airplane interior installation and has the following limitations as defined in document GC41548800 &quot;Guidance for Follow on Installation, Aft Galley/Galley Annex Multiple STC&quot;. A complete airplane installation...</td>
<td>10/02/2009</td>
</tr>
<tr>
<td>ST03100</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC SA0S-60</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Customized Interior Completion - AC5N 9293</td>
<td>TCCA STC No. SA05S-60 Issue 1 dated 14 June 2005. TCCA SOC Letter C-08-8674 dated 18 December 2008. Boeing Consulting Services Modification Data Summary 0329500 Revision BE dated 12 December 2008.</td>
<td>Restricted to ASIN 9293 and 1. Refer to the limitations sections of the required AFM Supplement and to the referenced FCOMS 2.2. WLAN system is compatible only to IEEE 802.11 b/g. Prior to installation of this modification the installer must determine that the installation and operation...</td>
<td>14/01/2009</td>
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<tr>
<td>ST03101</td>
<td>ELECTRONIC CABLE SPECIALISTS, Inc.</td>
<td>FAA STC ST05259CH</td>
<td>A321-213/211-311/215- 212/ A320-214/213/232/ A320-111/111-212-214/ A319-111-112-113-114- 115/ A330-111-112-113-114/ 115/ A318111-121-122</td>
<td>Installation of an EZMount™ Till Mechanical Mounting Assembly</td>
<td>None</td>
<td>Installation of an EZMount™ Till Mechanical Mounting Assembly</td>
<td>Prior to installation of this modification the installer must determine that the installation and any other previously installed modification will not introduce any adverse effect upon the airworthiness of the product. The installation of this modification is third person...</td>
</tr>
<tr>
<td>ST03102</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Installation, Complete Custom Aircraft Interior - ASN 8036</td>
<td>Transport Canada Limited STC No. C-LSA00-003/D dated 07 January 2007.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification is third person...</td>
<td>26/01/2009</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>EASA.IM.A.S.03104</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>ROSEMOUNT AEROSPACE, Inc.</td>
<td>BOEING MD 11F</td>
<td>Installation of Rosemount Aerospace 8700C-6 Series Laptop Docking Station (LDS) - Electronic Flight Bag (EFB) System Provisions in accordance with Rosemount Aerospace Master Data List, Doc. No. D09403165, Revision D, dated April 3, 2009 or later FAA approved revision. These provisions include Electr...</td>
<td>Airplane Flight Manual Supplement for MD-11 series AircraftRosemount Document No. D06403253 Rev. B</td>
<td>1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2) FAA approved Airplane Flight Manual Supplement, Rev...</td>
<td>10/06/2009</td>
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<tr>
<td>EASA.IM.A.S.03105</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>SA05-60</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>Transport Canada STC No. SA05-60 Issue 1 dated 14 June 2005, Transport Canada STC ref. C-09-0200 dated 31 March 2006, Aero Consulting Services Ltd. Configuration Definition List EQ25000 Revision BG or later approved revision Aero Consulting Services Ltd. Airplane Flight Manual Supplement C- FM25004501G...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third personnel...</td>
<td>02/04/2009</td>
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<td>EASA.IM.A.S.03117</td>
<td>BOMBARDIER AEROSPACE</td>
<td>FAA STC</td>
<td>TCCA C-LSA09-0002/D ISSUE 1</td>
<td>Main &amp; Nose Wheel Storage Modifications.</td>
<td>- TCCA STC No. C-LSA09-0002/D Issue 1 dated 19 January 2009. - Aero Consulting Services Modification Data Summary Number G150000 Revision NIC - Instructions for Continued Airworthiness G150070-ICA Revision N1 C2 or any later revisions of the above listed documents approved by EASA in accordance...</td>
<td>1. Restricted to ASN 5426.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of...</td>
<td>20/01/2009</td>
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<tr>
<td>EASA.IM.A.S.03124</td>
<td>ROSEN SUNVISOR SYSTEMS L.L.C.</td>
<td>FAA STC</td>
<td>LD41-500FG (C400)</td>
<td>Installation of Sunvisor.</td>
<td>According to Rosen Sunvisor System, LLC Master Drawing List No. 1381000-00LL, Rev. B or later FAA approved Revisions.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by th...</td>
<td>23/04/2009</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>STC ref.</td>
<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.IM.A.S.03125</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA C-LSA08-317/D</td>
<td>CL-600-2B16</td>
<td>Installation, Complete Custom Aircraft Interior</td>
<td>TCCA STC No C-LSA08-317/D Issue 1 dated 12 December 2006 Aero Consulting Services Ltd. Modification Data Summary GD48000 Revision NC or later approved revision Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation. 41-298M004 Revision NC or l.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>19/02/2009</td>
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<tr>
<td>EASA.IM.A.S.03126</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC SA01646SE</td>
<td>CESSNA A185F CESSNA A185E CESSNA A185D CESSNA A185C CESSNA A185B CESSNA A185A CESSNA A185 CESSNA A185B CESSNA 16U CESSNA 16H CESSNA 16D CESSNA 16C CESSNA 16B CESSNA 16A CESSNA 16 CESSNA 17S</td>
<td>Installation of vortex generators on the wings and tail surfaces in accordance with the provisions of AML SA01646SE.</td>
<td>Option 1: Installation of vortex generators on the wings, horizontal and vertical tail surfaces of the above modal Aircraft in accordance with Micro AeroDynamics Drawing Package MA2208, Revision IR, dated 17 January 2006 and Installation Manual MA2209, Revision IR, dated 27 January 2006 or later EAS...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>17/03/2009</td>
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<td>EASA.IM.A.S.03129</td>
<td>B/E AEROSPACE LIMITED</td>
<td>TCCA C-LSA09-022/D</td>
<td>CL-600-2B16</td>
<td>Installation, Complete Custom Aircraft Interior.</td>
<td>- TCCA STC No C-LSA09-022/D Issue 1 dated 03 April 2009. - Aero Consulting Services Ltd. Modification Data Summary G101500 Revision NC* - Midcoast Aviation Inc. Airplane Flight Manual Supplement for the Complete Custom Aircraft Interior Installation. 41-298M006 Rev NC* - Aero Consulting Sa...</td>
<td>1. Restricted to ASN 5734.2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 3. The installation...</td>
<td>14/04/2009</td>
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<tr>
<td>EASA.IM.A.S.03138</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC STA096A7-D</td>
<td>GULFSTREAM GV-SP</td>
<td>Installation of an Airplane Interior and Completion for Aircraft Serial5225.</td>
<td>- Install in accordance with Gulfstream Aerospace Corporation MasterIndex List, GCS1491S25 Rev C, dated May 05, 2009 or later FAA approved revision. - To be operated in accordance with FAA approved Airplane Flight Manual Supplement, ref. GCS1491M004, Rev. NC dated May 05, 2009, or later FAAapproved rev...</td>
<td>Aircraft Serial Number 5225 only.</td>
<td>20/05/2009</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>EASA.IM.A.S.03139</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>FAA STC ST0368AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of an Airplane Interior and completion for Aircraft Serial 4148.</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Master Index List, GCA14914148 Rev E dated March 17, 2009, or later FAA approved revision. To be operated in accordance with FAA approved Airplane Flight Manual Supplement, ref. GC41204M002, Rev. NC dated 03/17/09, or later FAA approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>07/04/2009</td>
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<td>EASA.IM.A.S.03140</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC SA06-34</td>
<td>BOMBARDIER BD-700-1A11</td>
<td>Installation of CMS-400 Digital Voice Checklist System.</td>
<td>- Transport Canada STC No. SA06-34 Issue 1 dated 23 May 2006. - Transport Canada SoC ref. C-09-0335 dated 15 April 2009. - Aero Consulting Services Ltd. Modification Data Summary no. ES938000 Revision A or later approved revision. - Bombardier Aerospace FMS No. G-FM31404001 Revision N/C or...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>27/04/2009</td>
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<td>EASA.IM.A.S.03141</td>
<td>BOMBARDIER COMPLETION CENTRE INC</td>
<td>TCCA STC SA04-113 ISSUE 1</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>Interior Completion, BD-700-1A10 - ASN 9281</td>
<td>Transport Canada SoC reference C-09-0174 dated 09 March 2009. TCCA STC No. SA04-113 Issue 1 dated 28 October 2004. Aero Consulting Services Configuration Definition List DA64500 Revision CB dated 11 December 2008. Instructions for Continued Airworthiness GC92810000-IFCA.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>10/03/2009</td>
</tr>
<tr>
<td>EASA.IM.A.S.03149</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>FAA STC ST02106LA-D</td>
<td>GV-SP (G550)</td>
<td>Installation of Custom Cabin Interior and Aircraft Completion for Aircraft Serial 5221.</td>
<td>Install in accordance with Gulfstream Aerospace Report GC520000653 Rev. NC and Index List GC520006643 Rev. C dated 20 April 2009, or later approved revisions. - Operate in accordance with Gulfstream Aerospace Flight Manual Supplement GC52000M006, revision N/ C, dated 20 April 2009, or later approved revision.</td>
<td>Aircraft Serial Number 5221 only</td>
<td>07/05/2009</td>
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<tr>
<td>EASA.IM.A.S.03153</td>
<td>ROSS FRESNO LILDBA CORPORATE AIRCRAFT</td>
<td>CESSNA 441 SERIES</td>
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<td>15/07/2009</td>
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<td>Date of Issue</td>
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<td>System Record</td>
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<td>STC4223-31</td>
<td>ROCKWELL COLLINS Inc.</td>
<td>AIRBUS A320-232</td>
<td>Installation of Sony Trans Com Inc. Video Entertainment System</td>
<td>• FAA STC SA0588NM- Sony Trans Com Inc. Index List, ref. doc. 062-0130, Revision AA, dated 12th May 2004 (<em>); Instruction for Continued Airworthiness, ref. doc. 070-428-000, Revision BF, dated 16th December 2005 (</em>); or later revisions approved by EASA in accordance with EASA ED decision...</td>
<td>29/04/2009</td>
<td>Active</td>
<td>10016670</td>
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<td>STC SA05-16</td>
<td>BOMBARDIER AEROSPACE</td>
<td>BD-100-1A10</td>
<td>Installation of KGS Electronic SS200 Static Inverter</td>
<td>• Transport Canada Supplemental Type Certificate SA09-16, Issue 2 dated 19th March 2009 in accordance with AeroConsulting Services Ltd. Modification Data Summary G072000 Revision B.</td>
<td>06/04/2009</td>
<td>Active</td>
<td>10016671</td>
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<td>STC SA0277CE</td>
<td>O&amp;N AIRCRAFT MODIFICATIONS, INC.</td>
<td>M20C, M20D, M20E, M20G</td>
<td>Installation of Fuel Bladder Cells in wet wing fuel cavities.</td>
<td>Installation and maintenance i.a.w. to: Configuration 1: Brentwing Engineering, Inc. Installation Drawing ListB-700 and/or O+i+k+i &gt;H Aircraft Modifications, Inc. Installation Instructions B-900 and/or O+i+k+i &gt;H Aircraft Modifications, Inc. Maintenance Instructions B-901, all in latest approved Revisions...</td>
<td>07/05/2009</td>
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<td>10043011</td>
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<tr>
<td>STC SA0277CE</td>
<td>O&amp;N AIRCRAFT MODIFICATIONS, INC.</td>
<td>M20F, M20J</td>
<td>Installation of Fuel Bladder Cells in wet wing fuel cavities.</td>
<td>Installation and maintenance i.a.w. to: Configuration 1: Brentwing Engineering, Inc. Installation Drawing ListB-400 and/or O+i+k+i &gt;H Aircraft Modifications, Inc. Installation Instructions Report B-300, all in latest approved Revisions...</td>
<td>07/05/2009</td>
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<td>EASA STC ref.</td>
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<td>FAA STC no.</td>
<td>Foreign/NA Aircraft Type Certif. no.</td>
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<td>Associated Technical Documentation</td>
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<td>A220</td>
<td>A.A.E., LTD.</td>
<td>FAA STC ST01077W</td>
<td></td>
<td>A220</td>
<td>Installation of AAE's smoke detection and fire suppression System in the lower Class C Cargo compartments.</td>
<td>- FAA STC ST01077W- Master Drawing List, ref. doc. TFA0-0057, Revision U, dated 11th March 2009.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by third person...</td>
</tr>
<tr>
<td>DHC-8-401, -402 DHC-8-301, -311, -314, -315 DHC-8-201, 202 DHC-8-120, -121, -103, -106</td>
<td>ROSEN SURVISOR SYSTEMS L.L.C.</td>
<td>FAA STC SA4339W</td>
<td></td>
<td>Summair Systems/FAA STC SA4339W dated October 31, 2008</td>
<td>Installation in accordance with Rosen Drawing List 201R0000, Revision 1 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>None</td>
<td>08/05/2009</td>
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<tr>
<td>SR20, SR22</td>
<td>AVIDYNE CORPORATION</td>
<td>FAA STC SA02913CH</td>
<td></td>
<td>Installation of Avidyne Model 9900EX/ TAS 600 Series Traffic Advisory System (TAS)</td>
<td>According to Avidyne Master Document List Doc. No. DC1029-1Rev. 12, or later FAA approved. Revisions to be operated in accordance with Avidyne AFM-S-Documents No. 32-23533C; Rev. 6 or later FAA approved. Revisions to be maintained i.e. a.w. ICA Document No. ICAW010807C, Rev. 12. or later approved. Revision...</td>
<td>27/05/2009</td>
<td>Active</td>
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<tr>
<td>BD-100-1A10</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCAC STC SA04-69 ISSUE 1</td>
<td></td>
<td>Installation of 115 VAC TIA Wavejet Microwave Oven in Forward RH Galley. EASA Validation of Transport Canada Supplemental Type Certificate SA04-69, Issue 1 dated 4th June 2004 in accordance with Aero Consulting Services Ltd. Modification Data Summary E027000 Revision N/C.</td>
<td>• Transport Canada Supplemental Type Certificate SA04-69, Issue 1 dated 4th June 2004.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by third person...</td>
<td>04/06/2009</td>
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<tr>
<td>DHC-8-314, DHC-8-315 DHC-8-301, DHC-8-311 DHC-8-201, DHC-8-202 DHC-8-812, DHC-8-103, DHC-8-106</td>
<td>BE AEROSPACE LIMITED</td>
<td>FAA STC SA91-97</td>
<td></td>
<td>Safety Catch Mechanism on the Rear Baggage Door/Sl. TCAC STC SA91-97, Issue 3, dated 18.11.2009. Note: This STC superseded EASA IM.A.S.03177.</td>
<td>Installation in accordance to ACS TCAC approved Modification Data Summary Report 956900, Rev. E, dated 28.06.2007. Maintenance in accordance to ACS Maintenance Manual Supplement, 956977, Rev. B, dated 08 November 2008.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by...</td>
<td>14/06/2009</td>
</tr>
<tr>
<td>DOA ST02219NY-D</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>FAA STC STO2219NY-D</td>
<td>BOEING 747-400 SERIES</td>
<td>Installation of Panasonic Avionics Corporation System 3000In-Flight Entertainment System in accordance with Delta Engineering Master Data List 0707-05814-2121 Rev. A dated Nov. 15, 2006 or later approved. Revision/Validation of FAA STC ST02219NY-D</td>
<td>Delta Engineering 0707-05814-2121 Rev A Instructions for Continued Airworthiness</td>
<td>1. Delta Engineering Instructions for Continued Airworthiness 0707-05814-2121 Rev. A dated Aug 31, 2006 are required with this installation. 2. The position of the seat installed in accordance with KLM LOPA KLM2112780 Revision ORG is required for this installation of the STC. The seats...</td>
<td>18/06/2009</td>
</tr>
<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.IM.A.S.03189</td>
<td>BOMBARDIER AEROSPACE</td>
<td>ST0283CH</td>
<td>BOEING 747-400 SERIES</td>
<td>Installation of EZMount (TM) mechanical mounting assembly on lower frame of cockpit sliding windows in accordance with Electronic Cable Specialists Master Data List ECS-206728 Rev. E, dated April 8, 2009, or later FAA approved revisions. Validation of FAA STC ST0283CH</td>
<td>AFM Supplement as listed in Appendix A of ECS Master Data List ECS-206728 Rev. E</td>
<td>1. AFM Supplement as listed in Appendix A of Electronic Cable Specialists Master Data List ECS-206728, Rev. E, dated April 8, 2009 or later FAA approved revision, is required on board the modified aircraft.2. The equipment for which these provisions are intended is not part of this approval. Additio...</td>
<td>18/06/2009</td>
</tr>
<tr>
<td>EASA.IM.A.S.03190</td>
<td>BAUMANN FLOATS LLC</td>
<td>SA1491CH</td>
<td>M-7-235B,M-7-235,MT-7-235,M-5-235C,M-6-235,M-7-235</td>
<td>Installation of BF2750A amphibious float installation</td>
<td>According to Baumann Master Document / Drawing List Doc. No. BF2750A-02Rev. F or later FAA approved revisions. To be operated in accordance with Baumann AFM-S Documents No. BF2150A, BF2550A &amp; B...</td>
<td>none</td>
<td>27/05/2009</td>
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<tr>
<td>EASA.IM.A.S.03193</td>
<td>ROCKWELL COLLINS AEROSPACE &amp; ELECTRONICS, Inc.</td>
<td>ST01295E</td>
<td>BOMBARDIER DHC-8-300 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-100 SERIES</td>
<td>Validation of the FAA STC ST001295E for the HGS 2800 on the Dash 8-100/200/300 Series Aircraft (editorial change).</td>
<td>German LBA STC (TA0092)Austrian AustroControl STC (ACG. 21NE.046)</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by th...</td>
<td>27/04/2009</td>
</tr>
<tr>
<td>EASA.IM.A.S.03195</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>ST01482NY</td>
<td>BOEING 747-400 SERIES</td>
<td>Installation of a Forward Galley-Vestibule Area-Forward Lavatory-Flight Deck Multiple STC</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Index List, GC421048504 Rev. D, dated January 30, 2009, or later FAA approved revision.</td>
<td>1. The Forward Galley- Vestibule Area-Forward Lavatory-Flight Deck Multiple STC, (FAA ref ST03649AT-D), does not approve the entire airplane completion and interior installation and has the following limitations as defined in document GC421041019, &quot;Guidance for Follow on Installation, Forward Galley-...</td>
<td>21/07/2009</td>
</tr>
<tr>
<td>EASA.IM.A.S.03196</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>ST01483NY</td>
<td>BOEING 747-400 SERIES</td>
<td>Installation of a Flight Deck-Forward Vestibule-Forward Lavatory Multiple STC</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Index List, GC421048504 Rev. D, dated January 30, 2009, or later FAA approved revision.</td>
<td>1. This Flight Deck-Forward Vestibule-Forward Lavatory Multiple STC, (FAA ref ST03649AT-D), does not approve the entire airplane completion and interior installation and has the following limitations as defined in document GC421041019, &quot;Guidance for Follow on Installation, Flight Deck-Forward Vestibul...</td>
<td>23/07/2009</td>
</tr>
<tr>
<td>EASA.IM.A.S.03197</td>
<td>DELTA ENGINEERING CORPORATION</td>
<td>ST01519NY</td>
<td>BOEING 747-400 SERIES</td>
<td>Installation of a Forward Galley-Vestibule Area-Forward Lavatory-Flight Deck Multiple STC</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Index List, GC421048504 Rev. D, dated January 30, 2009, or later FAA approved revision.</td>
<td>1. The Forward Galley- Vestibule Area-Forward Lavatory-Flight Deck Multiple STC, (FAA ref ST03649AT-D), does not approve the entire airplane completion and interior installation and has the following limitations as defined in document GC421041019, &quot;Guidance for Follow on Installation, Forward Galley-...</td>
<td>23/07/2009</td>
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<tr>
<td>EASA.IM.A.S.03203</td>
<td>GULFSTREAM AEROSPACE CORPORATION</td>
<td>ST03967AT-D</td>
<td>GIV-X</td>
<td>Installation of a Forward Galley-Vestibule Area-Forward Lavatory-Flight Deck Multiple STC</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Index List, GC421048504 Rev. D, dated January 30, 2009, or later FAA approved revision.</td>
<td>1. The Forward Galley- Vestibule Area-Forward Lavatory-Flight Deck Multiple STC, (FAA ref ST03649AT-D), does not approve the entire airplane completion and interior installation and has the following limitations as defined in document GC421041019, &quot;Guidance for Follow on Installation, Forward Galley-...</td>
<td>08/06/2009</td>
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<tr>
<td>EASA.IM.A.S.03204</td>
<td>GULFSTREAM SERVICES CORPORATION</td>
<td>ST03944AT-D</td>
<td>GULFSTREAM GIV-X</td>
<td>Installation of a Flight Deck-Forward Vestibule-Forward Lavatory Multiple STC</td>
<td>Install in accordance with Gulfstream Aerospace Corporation Index List, GC421048504 Rev. D, dated January 30, 2009, or later FAA approved revision.</td>
<td>1. This Flight Deck-Forward Vestibule-Forward Lavatory Multiple STC, (FAA ref ST03649AT-D), does not approve the entire airplane completion and interior installation and has the following limitations as defined in document GC421041019, &quot;Guidance for Follow on Installation, Flight Deck-Forward Vestibul...</td>
<td>10/06/2009</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. No.</td>
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<td>Associated Technical Documentation</td>
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<td>EASA.IM.A.S.03205</td>
<td>LEARJET Inc.</td>
<td>ST00604WI</td>
<td>LEARJET 31A</td>
<td>Installation of stand-alone Garmin GPS-165 Global Positioning System (VFR Navigation Only).</td>
<td>FAA STC ST00604WI with Master Document List, ref. doc. M100296ICT, Rev. B, dated 2 December 1997 or later revisions approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>• Airplane Flight Manual Supplement, ref. doc. AFMS-W1465, dated 1...</td>
<td>Prior to installation of this modification the installation must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
</tr>
<tr>
<td>EASA.IM.A.S.03220</td>
<td>AIRWELD, INC.</td>
<td>FAA STC ST00262BO</td>
<td>GULFSTREAM 200</td>
<td>Vapor Cycle Air Conditioning System Installation (Validation of FAA STC SA5008NM)</td>
<td>FAA STC SA5008NM, date reissued 10 April 2009 - World Auxiliary Power Company (WAPCO) Drawing List Number 506000, Revision 'A' dated 19 June 1990 or any later revision approved by EASA in accordance with EASA ED decision 2004/04/CF (or subsequent revisions of this decision).</td>
<td>The limitations of FAA STC SA5008NM apply</td>
<td>14/07/2009</td>
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<tr>
<td>EASA.IM.A.S.03231</td>
<td>PRATT &amp; WHITNEY ENGINE SERVICES, INC.</td>
<td>FAA STC ST00263BO</td>
<td>GULFSTREAM 200</td>
<td>Validation of FAA STC No. ST00263BO Data Transmission Unit (DTU) for Cessna Model 680</td>
<td>Certification Basis as per EASA TCDS IM.A.033 Issue 1 from 31 March 2005.</td>
<td>Master Document List DTU-G-100-4 Revision F from 23 July 2009, Instructions for Continued Airworthiness DTU-G-260-5 Revision B from 03 April 2009, Installation Instructions DTU-G-01-0-5 Revision B from 09 April 2009, or any other approved revision of this document</td>
<td>29/07/2009</td>
</tr>
<tr>
<td>EASA.IM.A.S.03233</td>
<td>MICRO AERODYNAMICS, INC.</td>
<td>FAA STC NO SA01555SE</td>
<td>A24, A24R, B24R, C24R, A23-19, A23-24, B23, C23 19A, B19, M19A, 23, A23, A23A</td>
<td>Installation of Vortex Generators on the wings and tail surfaces</td>
<td>FAA approved Drawing Package, Doc. No. MA2204, Rev. IR, dated 07- April-2005, or later approved revision of FAA approved installation manual, Doc. No. MA2205, Rev. IR, dated 04-May-2005, or later approved revision of FAA approved installation manual for Continued Airworthiness, signed by VG Operating Project, MAB801</td>
<td>1. Prior to installation of this modification the installation must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>02/07/2009</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>FAA/STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.IM.A.S.03234</td>
<td>CASCADE AVIATION SERVICES INC.</td>
<td>FAA ST16269SE</td>
<td>BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-100 SERIES</td>
<td>Installation of Cockpit Window Installation must be performed in accordance with Installation/Instructions Document R.S.01-2000 Rev. 0 or later revisions of the above-listed documents approved by EASA in accordance with EASA ED Decision2004/042/CF or subsequent revisions of this decision.</td>
<td>FAA STC ST01735LA-D</td>
<td></td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by third persons is to be done in accordance with Transport Canada approved Western Avionics Engineering Orders 71967-401 Revision 5 dated 17 September 2004 (91846-402 Revision 1 dated 17 september 2004).</td>
</tr>
<tr>
<td>EASA.IM.A.S.03247</td>
<td>BOMBARDIER AEROSPACE</td>
<td>TCCA STC SA07-110</td>
<td>CHALLENGER 300 BOMBARDIER BD-100-1A10</td>
<td>Installation of AVS-470 Tail Camera System. EASA Validation of Transport Canada Supplemental Type Certificate SA07-110; Issue 1 dated3rd August 2007 in accordance with Aero Consulting Services Ltd Modification Data Summary F161001.</td>
<td>FAA STC ST01735LA-D</td>
<td></td>
<td>02/07/2009</td>
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<tr>
<td>EASA.IM.A.S.03253</td>
<td>LEARJET INC.</td>
<td>FAA STC ST01735LA-O</td>
<td>BOMBARDIER CL-600-2B16 BOMBARDIER CL-600-2A12 BOMBARDIER CL-600-1A11</td>
<td>FAA STC ST01735LA-D; Installation of a Kidde Smoke Detector System in the Aft Baggage Compartment, as a replacement for Walter Kidde PIN473052.</td>
<td>FAA STC No. ST01735LA-D dated 13 March 2007 BOMBARDIER Aerospace Master Document List BA69001 0115 Revision A dated13 March 2007 or later approved revision. BOMBARDIER Aerospace Airplane Flight Manual Supplement BA69021SF44 Revision NC dated 13 March 2007 or later approved revision is required for the installation of this STC.</td>
<td>An EASA approved Walter Kidde Smoke Detector System for PIN473052 is a prerequisite for this STC.</td>
<td>02/07/2009</td>
</tr>
<tr>
<td>EASA.IM.A.S.C.01002</td>
<td>WESTERN AVIONICS</td>
<td>TCCA STC SA03-13; EASA.IM.A.S. BOMBARDIER DHC-8-315 BOMBARDIER DHC-8-311 BOMBARDIER DHC-8-300 SERIES BOMBARDIER DHC-8-200 SERIES BOMBARDIER DHC-8-103 BOMBARDIER DHC-8-102 BOMBARDIER DHC-8-100 SERIES</td>
<td>Installation (Upgrade) of a Single or Dual Universal Avionics UNS-1+ or UNS-1E Flight Management System (FMS)</td>
<td>- Top Document List 91846-400, Revision 5 dated 21 September 2004 Western Avionics AFM Supplement 71967-401, Revision 1 dated 05 August 2003 (EM) or 71967-804, Initial Issue dated05 August 2003 (EFIS), (AFM Supplement 71967, Revision 1 dated 05 August 2003 (EM) of AFM Supplement 71967-804, India...</td>
<td>1. Installation (Upgrade) of the Single or Dual Universal Avionics UNS-1+ or UNS-1E Flight Management System (FMS) is to be done in accordance with Transport Canada approved Western Avionics Engineering Orders 71967-401 Revision 5 dated 17 September 2004 (91846-402 Revision 1 dated 17 september 2004)...</td>
<td>20/05/2005</td>
<td>Active</td>
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<td>STC No.</td>
<td>STC Holder</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>ST01443LA</td>
<td>AIRWORKS</td>
<td>BOEING 737-900 SERIES, BOEING 737-600/700/800, BOEING 737-300/400-500</td>
<td>Installation of AirWorks Cockpit Door Surveillance Systems</td>
<td>- Aircraft Master Data List MDL23006, Revision A dated 11/11/2004 or later FAA approved Revision AirWorks Airliner Flight Manual Supplement Document No. AFM230006, Revision Original dated 11/02/2005 or later FAA approved revision.</td>
<td>FAA STC No. ST01443LA amended 3/2005.2</td>
<td>1. Limitations according to FAA STC No. ST01443LA amended 3/2005.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>03/08/2005</td>
</tr>
<tr>
<td>SB00463AT</td>
<td>HEAD BALLOONS, INC.</td>
<td>HEAD BALLOONS AX9-118, HEAD BALLOONS AX8-88, HEAD BALLOONS AX8-105, HEAD BALLOONS AX7-77B, HEAD BALLOONS AX7-77</td>
<td>Installation of Cameron Balloons Ltd. baskets and burners underneath Head Balloons, Inc. envelopes</td>
<td>Balloon Flight Manual (BFM) Supplements per model. BFM supplement for Head Balloons AX7-77 w/Parachute top, revision 1, dated OCT 21 2005 – BFM supplement for Head Balloons AX7-77b w/Parachute top, revision 1, dated OCT 21 2005 – BFM supplement for Head Balloons AX8-88 w/Parachute top, rev...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The Cameron Balloons Ltd. baskets and burner...</td>
<td>19/11/2008</td>
<td>Active</td>
</tr>
<tr>
<td>IO-550-F</td>
<td>TEXAS SKYWAYS, INC.</td>
<td></td>
<td>Removal of fuel injection system and induction system and installation of carburetor in accordance with Texas Skyways, Inc. Specification TS0030, dated 01 May 1995 or FAA later approved revision, based on FAA Supplemental Type Certificate Number SE091385C</td>
<td>Texas Skyways, Inc. Specification TS0030, dated 01 May 1995 or later approved revision.</td>
<td>Approval of this change in type design applies to the above engine only. Compatibility of this modification with previously approved modifications must be determined by the installer. A copy of this Certificate must be maintained as part of the permanent records for the modified engine. The...</td>
<td>06/04/2006</td>
<td>Active</td>
</tr>
<tr>
<td>REV 1</td>
<td>AVIATION FLEET SOLUTIONS, LLC</td>
<td>JTBD-217, JTBD-21TA, JTBD-217C, JTBD-219</td>
<td>Supplemental Noise Configuration: noise addition of normal TO Thrust + Automatic Reserve Thrust System and Landing Flap28. The previous FAA STC SE01650SE dated 21.03.2006 was approved under EASA approval EASA.IM.E.S.01001 dated 8 June 2006.</td>
<td>FAA approved AFM Supplement AFMS-MD80-1 Rev. A, dated 3 August 2006 for Airplane models MD-81/82/83/85 FAA approved AFM Supplement AFMS-MD83-7, Rev. A, dated 3 August 2006 for Airplane models MD-87 FAA approved AFM Supplement AFMS-MD80-9 (normal TO + ARTS), Rev. IR, dated 3 August 2006 for Airp...</td>
<td>1. See above AFM Supplement for this STC is approved only for the product configuration as defined in the STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall be determined by the...</td>
<td>12/02/2007</td>
<td>Active</td>
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<tr>
<td>EASA STC no.</td>
<td>FAA STC</td>
<td>Foreign/NAA STC no.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>EASA.IM.E.S.01002</td>
<td>PETERSEN AVIATION, INC.</td>
<td>FAA STC</td>
<td>SE00639W</td>
<td>LYCOMING ENGINES O-540-F1B5</td>
<td>Adhesive following approved fuel. Leaded and unleaded automotive gasoline, 91 minimum antiknock index (RON+MON)/2 per ASTM Specification D-4390-D-4814 or per European Norm Standard EN 228. Interchanging with aviation gasoline is also approved.</td>
<td>Installation of modified Oil Filter Adapter, an increased opening pressure spring for filter bypass, and modified filter stud in accordance with Firewall Forward Technologies, Inc. Master Document List No. FFT-MA000001, Rev. B, dated August 8, 2005 or later FAA approved revision.</td>
<td>1. This STC is limited to engine installations on Robinson Model R44 Helicopters only. The manifold pressure and RPM limitations in the FAA approved Robinson R44 Flight Manual must be maintained.</td>
</tr>
<tr>
<td>EASA.IM.E.S.01005</td>
<td>CENTRILUBE TECHNOLOGIES, LLC</td>
<td>FAA STC</td>
<td>SE00534E</td>
<td>TIO541A,-E SE., TIO541E SE.</td>
<td>Installation of modified Oil Filter Adapter, an increased opening pressure spring for filter bypass, and modified filter stud in accordance with Firewall Forward Technologies, Inc. Master Document List No. FFT-MA000001, Rev. B, dated August 8, 2005 or later FAA approved revision.</td>
<td>Installation Instructions FFT-MLYLC-OLSTUD-001 Continued Airworthiness Instructions FFT-CAILYLC-OLSTUD-001</td>
<td>Compatibility of this design change with previously approved modifications must be determined by the installer if the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission. Refer to original equivalent.</td>
</tr>
<tr>
<td>EASA.IM.E.S.01006</td>
<td>PRATT &amp; WHITNEY</td>
<td>FAA STC</td>
<td>SE00033EN</td>
<td>CFM56-3B1</td>
<td>Installation of Pratt &amp; Whitney (P&amp;W) Global Material Solutions (GMS) CFM56-3 High Pressure Compressor Module in accordance with P&amp;W GMS Drawing List 88500-01C dated May 06, 2008 or later FAA approved revision EASA Certification Basis: 1. Airworthiness Standards JAR-E Change 2 (BCAR Issue 9 dated 16...</td>
<td>Refer to P&amp;W GMS Service Bulletin PWGMS 72-2 Revision 2, dated December 17, 2008, and later approved revision.</td>
<td>Limitations 1. The limitations of the EASA Engine Type Certificate Data Sheet E.066 issue 1, dated November 28, 2008 apply to this Supplemental Type Certificate. Life limits for the parts approved by this Supplemental Type Certificate are published in the Airworthiness Limitations section of...</td>
</tr>
<tr>
<td>EASA.IM.E.S.01007</td>
<td>PRATT &amp; WHITNEY</td>
<td>FAA STC</td>
<td>SE00034EN</td>
<td>CFM56-3B1</td>
<td>Installation of Pratt &amp; Whitney (P&amp;W) Global Material Solutions (GMS) CFM56-3 High Pressure Turbine Module in accordance with P&amp;W GMS Drawing List 88500-01D dated May 06, 2008 or later FAA approved revision EASA Certification Basis: 1. Airworthiness Standards JAR-E Change 2 (BCAR Issue 9 dated 18...</td>
<td>Refer to P&amp;W GMS Service Bulletin PWGMS 73-3 Revision 2, dated December 17, 2008, and later approved revision.</td>
<td>Limitations 1. The limitations of the EASA Engine Type Certificate Data Sheet E.066 issue 1, dated November 28, 2008 apply to this Supplemental Type Certificate. Life limits for the parts approved by this Supplemental Type Certificate are published in the Airworthiness Limitations section of...</td>
</tr>
<tr>
<td>EASA.IM.E.S.01008</td>
<td>PRATT &amp; WHITNEY</td>
<td>FAA STC</td>
<td>SE00035EN</td>
<td>CFM56-3B1</td>
<td>Installation of Pratt &amp; Whitney (P&amp;W) Global Material Solutions (GMS) CFM56-3 Low Pressure Turbine Module in accordance with P&amp;W GMS Drawing List 88500-01E dated May 06, 2008 or later FAA approved revision EASA Certification Basis: 1. Airworthiness Standards JAR-E Change 2 (BCAR Issue 9 dated 18...</td>
<td>Refer to P&amp;W GMS Service Bulletin PWGMS 74-3 Revision 2, dated December 17, 2008, and later approved revision.</td>
<td>Limitations 1. The limitations of the EASA Engine Type Certificate Data Sheet E.066 issue 1, dated November 28, 2008 apply to this Supplemental Type Certificate. Life limits for the parts approved by this Supplemental Type Certificate are published in the Airworthiness Limitations section of...</td>
</tr>
<tr>
<td>EASA.IM.E.S.01009</td>
<td>PRATT &amp; WHITNEY</td>
<td>FAA STC</td>
<td>SE00032EN</td>
<td>CFM56-3B1</td>
<td>Installation of Pratt &amp; Whitney (P&amp;W) Global Material Solutions (GMS) CFM56-3 Fan and Booster Module in accordance with P&amp;W GMS Drawing List 88500-01F dated March 14, 2008 or later FAA approved revision EASA Certification Basis: 1. Airworthiness Standards JAR-E Change 2 (BCAR Issue 9 dated 18...</td>
<td>Refer to P&amp;W GMS Service Bulletin PWGMS 72-1 Revision 2, dated December 17, 2008, and later approved revision.</td>
<td>Limitations 1. The limitations of the EASA Engine Type Certificate Data Sheet E.066 issue 1, dated November 28, 2008 apply to this Supplemental Type Certificate. Life limits for the parts approved by this Supplemental Type Certificate are published in the Airworthiness Limitations section of...</td>
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**List of EASA Supplemental Type Certificates**

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<th>STC No.</th>
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<th>Model</th>
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<th>Certification basis</th>
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<tr>
<td>EASA.IM.E.S.01010</td>
<td>CENTRI-LUBE TECHNOLOGIES, LLC</td>
<td>FAA STC SE01436LA</td>
<td>TIO-541A, E-5E, TIO-541A-540, TIO-540-540</td>
<td>Installation of Firewall Forward Technologies, LLC modifications in accordance with Master Drawing List MWL-ST7012A(E), revision G, dated 2/14/03 or later FAA approved revision.</td>
<td>Installation Instructions FT-16-001 Continued Airworthiness Instructions FT-CAI-001</td>
<td>04/02/2007</td>
<td>Active</td>
<td>10116633</td>
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<tr>
<td>EASA.IM.E.S.01011</td>
<td>CENTRI-LUBE TECHNOLOGIES, LLC</td>
<td>REV. 1</td>
<td>FAA STC SE00590E</td>
<td>SEE FAA AML SE00590E/05/00/160/15-160/10-10/00/10-0</td>
<td>Installation of a Lubricated Camshaft in accordance with Master Drawing List FT-ML-ST3370D/E, Revision IR dated September 9, 2005, FAA approved September 26, 2005, or later FAA approved revision.</td>
<td>Installation Instructions FT-16-002 Continued Airworthiness Instructions FT-CAI-002</td>
<td>05/06/2007</td>
<td>Active</td>
<td>10116697</td>
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<tr>
<td>EASA.IM.E.S.01016</td>
<td>AERODYNE CORPORATION</td>
<td>FAA STC SE0024280</td>
<td>IO-580-B1A</td>
<td>Installation of induction intake pipes in accordance with Aerodyne 114/A8/STC 320 HP Engine Conversion Installation Instructions, dated July 20th 2006 or later revisions of the above listed document approved by EASA in accordance with EIC decision 2004/04/CF (or subsequent reversion of this decision).</td>
<td>Instructions for operation, maintenance and inspection Doc.No. AD526-01: 114/A8/STC 320HP Engine Conversion – Instructions for Continued Airworthiness</td>
<td>04/03/2008</td>
<td>Active</td>
<td>10116701</td>
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<tr>
<td>EASA.IM.E.S.01017</td>
<td>PRATT &amp; WHITNEY</td>
<td>FAA STC SE00326EN</td>
<td>ROLLS-ROYCE 250-C20 S/2</td>
<td>Installation of Pratt &amp; Whitney Engine Services, Inc. IntellStart Plus +TM Electronic Control Unit (ECU), PN DPU- K-010 on the specified 250 engine models.</td>
<td>Airframe installation of the Allied Aerospace Corporation - AlliedSignal Aerospace IntellStart Plus+TM to be accomplished in accordance with applicable airframe STC.</td>
<td>14/03/2008</td>
<td>Active</td>
<td>10116702</td>
<td></td>
</tr>
<tr>
<td>EASA.IM.E.S.01018</td>
<td>PRATT &amp; WHITNEY</td>
<td>FAA STC SE00325EN, LBA EN003A</td>
<td>HONEYWELL INT. TS3/138</td>
<td>Installation of Pratt &amp; Whitney Engine Services, Inc. SmartCycle Plus Engine Monitoring System, PN EMU-K-010-1 on the specified TS3 engine models</td>
<td>1. SmartCycle Plus Engine Monitoring/Display is designed as an electronic aid for recording engine trend and cyclic data. This system does not eliminate any scheduled maintenance or parts replacement based on time or any engine Instructions for Continued Airworthiness monitoring requirements. Nor does i...</td>
<td>04/03/2008</td>
<td>Active</td>
<td>10116703</td>
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### List of EASA Supplemental Type Certificates

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<th>STC No.</th>
<th>STC Holder</th>
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<th>Certification basis</th>
<th>Installation Limitations/Conditions</th>
<th>Date of Issue</th>
<th>Status</th>
<th>System Record</th>
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<tr>
<td>EASA.IM.E.S.0010</td>
<td>PRATT &amp; WHITNEY</td>
<td>BELL 412 EP</td>
<td>Installation of Pratt &amp; Whitney Engine Services, Inc. IntelliStart Plus +TM Electronic Control Unit (ECU), PN DPU-K-010 on the specified engine models</td>
<td>1. The IntelliStart Plus +TM ECU is designed to limit start temperatures and to provide an electronic aid for recording engine trend and cycle data.2. The IntelliStart Plus +TM ECU is designed as non-essential, ground use only equipment and cannot be used to replace pilot monitoring during start as...</td>
<td>14/03/2008</td>
<td>Active</td>
<td>10118199</td>
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<tr>
<td>EASA.IM.P.S.01003</td>
<td>WESTERN PROPELLER (PACIFIC) LTD.</td>
<td>TCCA STC SP96-1</td>
<td>Installation of a Zero Stop Latch Assembly to hold the propeller in zero pitch during engine start to prevent float and site equipped aircraft from surging forward during transition from feather to beta range. In accordance with the data contained in Western Propeller Master Document List No. WPL-C703...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this...</td>
<td>23/01/2009</td>
<td>Active</td>
<td>10018135</td>
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<tr>
<td>EASA.IM.R.S.0030</td>
<td>INTEC HELICOPTERS Inc.</td>
<td>BELL 407 BELL 206 A,B,B1, L-1, L-3, L-4</td>
<td>Installation of Engine air filter in accordance with INTEC Master Drawing List No. 1206 No. 1006-1001 Revision F1, dated 30 JAN 2004 (or later EASA Approved revision).</td>
<td>1. A copy of this STC and the appropriate approved Flight Manual Supplement (see INTEC Master Drawing List No. &quot;1006-1001&quot;) must be maintained as part of the permanent records for the modified aircraft.Additionally, for Bell Model 206-A, B and B1 aircraft, a copy of Bell Helicopter Flight Manual Supplement...</td>
<td>09/07/2004</td>
<td>Active</td>
<td>10043065</td>
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<tr>
<td>EASA.IM.R.S.0039</td>
<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>MD900</td>
<td>Installation of Apical Industries, Inc. Emergency Floats and 6-man LifeRaft Kits, Part Number 20754-100, or -200, in accordance with 9000-2. Installation Instructions MD900 Emergency Float Kits, Revision E, dated 1 April 2002 or later EASA-approved revision.</td>
<td>1. This approval requires the inspections and overhaul schedules to be performed as mandated by the Instructions for Continued Airworthiness, ICA 900-2, dated 25 July 2001, or later EASA approved revision.2. The Rotorcraft Flight Manual Supplement FMS900-3 is required when operating with the...</td>
<td>13/07/2004</td>
<td>Active</td>
<td>10261996</td>
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<tr>
<td>EASA.IM.R.S.0048</td>
<td>PRATT &amp; WHITNEY</td>
<td>REV.1</td>
<td>BELL 412 EP BELL 412</td>
<td>Installation of Pratt &amp; Whitney Engine Services, Inc. SmartCycle monitoring and recording system</td>
<td>1. Required maintenance, as prescribed in the applicable airframe or engine Instructions for Continued Airworthiness, shall not be predicted on the data obtained from Pratt &amp; Whitney Engine Services, Inc., monitoring/recording system...</td>
<td>25/02/2005</td>
<td>Active</td>
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<tr>
<td>EASA.IM.R.S.00548</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>427</td>
<td>Manufacture and Installation of Passenger Sliding Door for Bell 427 Model Helicopters (FAA STC SR02381AT dated 19/03/2002)</td>
<td>1. Rotorcraft Flight Manual Supplement dated 7/06/2001, is a required part of this STC.2. &quot;Instruction for Continued Airworthiness&quot; AA Report No. AA-99043, Revision A dated 6/10/2000 is also a required part of this STC.3. This STC is approved only for the product configuration as...</td>
<td>26/01/2005</td>
<td>Active</td>
<td>10161810</td>
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<td>EASA STC No.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC No.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<tr>
<td>EASA.IM.R.S.00550</td>
<td>AERONAUTICAL ACCESSORIES, INC</td>
<td></td>
<td>FAA STC SR02190AT</td>
<td>BELL MODEL 427</td>
<td>Manufacture and Installation of Floatstep® Floatstep for Bell 427 Model Helicopters (FAA STC SR02190AT)</td>
<td>- A41 Internal Drawing List Report AA-08149 dated 31/08/2000. - Installation instructions to Floatstep Kit P/N 427-303-001.</td>
<td>1. Inspection and Maintenance Requirements are listed in AeronauticalAccessories, Inc. Report No. AA-09048, revision B dated 23/02/2000 or later FAA approved revision. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
</tr>
<tr>
<td>EASA.IM.R.S.00619</td>
<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td></td>
<td>FAA STC SR00470LA</td>
<td>AS 350 B/BA/B1/B2/B3</td>
<td>Emergency Floatation Kit installation according to FAA STC No. SR00470LA issued on November 17, 1997 and reissued December 22, 1998.</td>
<td>- EASA-approved AS 350 Flight Manual and RFM Supplement Aipal FMS-350(dated October 15, 1997, or later FAA approved revisions) are required.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<tr>
<td>EASA.IM.R.S.00623</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td></td>
<td>FAA STC SR01186SE</td>
<td>EUROCOPTER FRANCE AS 350 B3</td>
<td>Cargo Hook Kit installation, in accordance with FAA STC No SR01186SE issued January 22, 2003.</td>
<td>- AS 350 B3 Flight Manual and RFM Supplement Onboard No. 121-014-00 dated January 15, 2003, and revised on July 23, 2004 or later approved revisions.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<tr>
<td>EASA.IM.R.S.00624</td>
<td>INTEC HELICOPTERS Inc.</td>
<td></td>
<td>EC 120 B</td>
<td></td>
<td>Installation of an Engine Air Filter, in accordance with FAA STC No SR91190SE, issued on March 17, 2003</td>
<td></td>
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<tr>
<td>EASA.IM.R.S.00709</td>
<td>EAGLE COPTERS MAINTENANCE Ltd.</td>
<td></td>
<td>TCCA STC No. SH92-43</td>
<td>412, 412 CF, 412 EP 212, 216B, 216B-1 204, 205A-1, 205B</td>
<td>TC STC SH93-43 Serviceable Kit Auxiliary Crew. Latches, Cockpit Door/Support Kit, Cabin Door Handles and Modified Roller System for sliding cabin door (configuration 2)</td>
<td>Dart Aerospace Ltd. Installation Instruction INNI-D412-698 Rev. B or later approved revision.</td>
<td>1. This approval is limited to configuration 2 only. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
</tr>
<tr>
<td>EASA.IM.R.S.00710</td>
<td>DART AEROSPACE Ltd.</td>
<td>REV. 1</td>
<td>TCCA STC SH04-7</td>
<td>BELL 412 EP BELL 412 BELL 214B BELL 212 BELL 205B BELL 205A</td>
<td>Installation of D412-702 Shoulder Harness Kits. Transport Canada STC No SH94-7 issued 6th February 2004. The purpose of these kits is to add shoulder harnesses to existing seats that are only equipped with lap belts.</td>
<td>Prior to installation of this modification the installer must determine that the relationship between this modification and any other previously installed modifications will introduce no adverse affect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>09/08/2007</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev.</td>
<td>Foreign/NA</td>
<td>STC no.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
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<tr>
<td>EASA.IM.R.S.00712</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCIA STC</td>
<td>NO. SH04-25</td>
<td>212, 412, 412 CF, 412 EF</td>
<td>Installation of Twin Park E-Z Access Kit - STC SH04-25</td>
<td>Dart Aerospace Ltd. Installation instruction INN-D412-705 Rev. A dated February 20, 2004 or later Transport Canada approved revision/Instruction/Continued Airworthiness ICA-D412-705, Initial issue dated March 9, 2004 or later revision are required with this installation.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
</tr>
<tr>
<td>EASA.IM.R.S.00713</td>
<td>HELIPOD USA LLC</td>
<td>R44, R44II</td>
<td>validation of FAA STC SR01418SE for the Single or Dual Installation of External Cargo Pods on Robinson R44 and R44II.</td>
<td>Master Drawing List MLD. NW011-HP-25-005 Rev. 1 dated 02.05.2004 or later EASA approved revisions/Installation Instructions/NW011-HP-25-005 dated 05.02.2004 or later EASA approved revisions in situations for Continued Airworthiness NW011-HP-25-005 dated 03.05.2004 or later EASA approved revisions FAA appr...</td>
<td>1. The installation of the Helipod External Cargo Pods is compatible with any float installations. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>27/06/2005</td>
<td>Active</td>
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<tr>
<td>EASA.IM.R.S.00718</td>
<td>SIMPLEX MANUFACTURING CO.</td>
<td>FAA STC</td>
<td>SR00839NY</td>
<td>EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B3</td>
<td>Installation of Simplex anti-fire system Model 310-600, 310-1100 and 310-1200</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/10/2004</td>
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<tr>
<td>EASA.IM.R.S.01011</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC</td>
<td>SR01593NY-D</td>
<td>SIKORSKY S-76C SIKORSKY S-76B SIKORSKY S-76A</td>
<td>Installation of Luminescent Systems.</td>
<td>- FAA STC SR01133NY-D - Keystone Helicopter Instructions for Continued Airworthiness Report No HP-25-005 dated April 12, 2002 or later approved revision. - R.P.M. Supplement No FM00565, no revision, approved December 10, 2002</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.IM.R.S.01012</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC</td>
<td>SR01593NY-D</td>
<td>SIKORSKY S-76C SIKORSKY S-76B SIKORSKY S-76A</td>
<td>Installation of Keystone Helicopter Utility Interior Panels</td>
<td>- FAA STC No SR01593NY-D - Keystone Helicopter Instructions for Continued Airworthiness Report No IC00735 Rev. dated June 11, 2004 or later approved revision. 2. It is a prerequisite to have previously approved brackets installed for each individual panel installation as denoted in Keystone Helicopter IM00569, Installation 3.</td>
<td>22/04/2005</td>
</tr>
<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>EASA.IM.R.S.01013</td>
<td>APICAL INDUSTRIES, Inc.</td>
<td>FAA STC SR00760LA</td>
<td></td>
<td>EC 120 B</td>
<td>Installation of Apical Emergency Float Kit, p/n 20498. The present STC is the validation of the FAA STC n° SR00760LA issued on 19/03/1999.</td>
<td>FAA STC No SR00760LA issued on 19/03/1999 – Definition: Master Document List MD120-1, revision D dated March 18, 1999 approved by FAA or later FAA-approved revisions. Installation, Installation and Packing Instructions I120-100, revision C dated May 12, 1999 approved by FAA or later FAA-approved.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<tr>
<td>EASA.IM.R.S.01014</td>
<td>AEROLITE AG</td>
<td>-</td>
<td>A109E</td>
<td>A109E</td>
<td>Installation of Medical Oxygen Cabinet P/ N 109111-501</td>
<td>-Installation Manual IM-502</td>
<td>1. No Smoking when Medical Oxygen Cabinet is installed.</td>
</tr>
<tr>
<td>EASA.IM.R.S.01016</td>
<td>GEOTECH Ltd.</td>
<td>-</td>
<td>AS 350 SERIE</td>
<td>Survey System Installation (Structural &amp; Electrical Power provisions)</td>
<td></td>
<td></td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
</tr>
<tr>
<td>EASA.IM.R.S.01018</td>
<td>ISOLAR Inc., ISOLAR HELICOPTER BY</td>
<td>FAA STC SR01410NY</td>
<td>AS350 B1, -B2, -B3, -BA</td>
<td>Installation of Isolair spray system model 3900-350B2.</td>
<td></td>
<td></td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by third person...</td>
</tr>
<tr>
<td>EASA.IM.R.S.01019</td>
<td>AEROLITE AG</td>
<td>-</td>
<td>MD900, MD902</td>
<td>External Hold Mirror Installation P/N 9005524-501, P/N 9005524-502, P/N 9005524-503, P/N 9005524-5A Flight Manual Supplement FMS-404.</td>
<td></td>
<td></td>
<td>1. Only for VFR operations. 2. Only approved configurations to be installed. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the...</td>
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<tr>
<td>EASA.IM.R.S.01021</td>
<td>DART AEROSPACE Ltd.</td>
<td>SH04-31</td>
<td></td>
<td>Installation of Replacement Window Kits</td>
<td>Installation must be in accordance with Dart Aerospace Installation Instruction SN-D412-709rev A or later approved revision Maintenance must be in accordance with Dart Aerospace Instruction for Continued Airworthiness ICA-D412-709 Rev A. TCCA accepted or later approved revision</td>
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<td>02/01/2006</td>
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<td>STC Holder</td>
<td>STC No.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
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<td>Status</td>
<td>System Record</td>
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<tr>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>STC SR01132NY-D</td>
<td>Installation of Honeywell KD550 Multifunctional Display (MFD) displays on a Honeywell KD550 Multifunctional Display (MFD)</td>
<td>FAA STC SR01132NY-D</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft engine configurations shall be determined by the installer.</td>
<td>22/04/2006</td>
<td>Active</td>
<td>10101734</td>
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<tr>
<td>TECH-TOOL PLASTICS, Inc.</td>
<td>STC SR00033NY</td>
<td>Installation of Grease Fittings.</td>
<td>FAA STC SR00033NY</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>22/11/2005</td>
<td>Active</td>
<td>10101730</td>
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<tr>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>STC SR01596NY-D</td>
<td>Installation of Grease Fittings.</td>
<td>FAA STC SR01596NY-D</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/03/2006</td>
<td>Active</td>
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<tr>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>STC SR01596NY-D</td>
<td>Installation of Grease Fittings.</td>
<td>FAA STC SR01596NY-D</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>18/11/2005</td>
<td>Active</td>
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**List of EASA Supplemental Type Certificates**

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<th>STC Holder</th>
<th>STC No.</th>
<th>Description</th>
<th>Certification basis</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<th>System Record</th>
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<tbody>
<tr>
<td>EASA.IM.R.S.01022</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>Installation of Honeywell KD550 Multifunctional Display (MFD) displays on a Honeywell KD550 Multifunctional Display (MFD)</td>
<td>FAA STC SR01132NY-D</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft engine configurations shall be determined by the installer.</td>
<td>22/04/2006</td>
<td>Active</td>
<td>10101734</td>
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<tr>
<td>EASA.IM.R.S.01028</td>
<td>HERITAGE AVIATION LIMITED</td>
<td>Installation of Grease Fittings.</td>
<td>FAA STC SR00033NY</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>22/11/2005</td>
<td>Active</td>
<td>10101730</td>
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<tr>
<td>EASA.IM.R.S.01032</td>
<td>TECH-TOOL PLASTICS, Inc.</td>
<td>Installation of Grease Fittings.</td>
<td>FAA STC SR00033NY</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>06/03/2006</td>
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<td>EASA STC No.</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
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<td>EASA.IM.R.S.01060</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td></td>
<td>FAA STC SR01638NY-D</td>
<td>SIKORSKY S-76C</td>
<td>Installation of Aircraft ST 3100 Satellite Telephone System</td>
<td>Keystone Helicopter installation drawing list IDU03783, Revision A dated 5/18/04 or later approved revision.</td>
<td>Keystone Helicopter AFM Supplement No FM0783 Rev 1 dated 24 August 2005 or later approved revision</td>
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<tr>
<td>EASA.IM.R.S.01063</td>
<td>TEST INSTRUMENTATION INC.</td>
<td></td>
<td>FAA STC SR00052SE</td>
<td>R44, R44 11</td>
<td>Installation of Classic Aircraft products Cargo Hook System-801 in accordance with FAA STC SR00052SE</td>
<td>EASA approved Rotorcraft Flight Manual No 44-510EU</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
</tr>
<tr>
<td>EASA.IM.R.S.01064</td>
<td>AEROMETALS INC.</td>
<td></td>
<td>FAA STC SH0736NN</td>
<td>MD HELICOPTERS 369H, 369HE, 369HS</td>
<td>Installation of Aerometals tailboom P/N 369X3500-505 of MDHI tailboom P/N YH-203. The present STC is the validation of the FAA STC SH3576NN amended on 30/11/99, limited to MDHI 369H, HS and HE models.</td>
<td>Aerometals Tailboom P/N 369X3500-505 - Installation Instructions: Aerometals Master Drawing List (MDL) No 25, revision B dated 22/10/99 approved by FAA or later FAA-approved revision. - Instructions for Continued Airworthiness: Aerometals Instructions for Continued Airworthiness AMS-14 dated 04/05/99.</td>
<td>1. Service life of the Aerometals and Yehnert referenced tailbooms is established according to FAR 27.571 (e) at 10000 flight hours, with periodic inspections required at 2450 flight hours and each 2450 flight hours thereafter. 2. Inspections of the Aerometals and Yehnert referenced tailbooms shall be conducted in...</td>
</tr>
<tr>
<td>EASA.IM.R.S.01065</td>
<td>AEROMETALS INC.</td>
<td></td>
<td>FAA STC SH0055NN</td>
<td>369D, 369E, 369FF</td>
<td>Installation of Aerometals tailboom P/N 369X23500-505 of MDHI tailboom P/N 369X23500-505 or YH-225-1 as replacement of MDHI tailboom P/N 369X23500-101, or P/N YH-225-3 as replacement of YH-225-1.</td>
<td>Installation Instructions and Instructions for Continued Airworthiness; see table in the paragraph &quot;Limitations and Conditions&quot; hereafter.</td>
<td>1. Service life of the Aerometals and Yehnert referenced tailbooms is established according to FAR 27.571 (e) at 10300 flight hours with periodic inspections required at 7300 flight hours and each 1000 flight hours thereafter. 2. Inspections of the Aerometals and Yehnert referenced tailbooms shall be conducted in...</td>
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<tr>
<td>EASA.IM.R.S.01069</td>
<td>DEVORE AVIATION CORPORATION OF AMERICA</td>
<td></td>
<td>FAA STC SH4694SW</td>
<td>S-76A/B/C</td>
<td>Installation of Under Fuselage forward facing recognition lights and rotor disc lights</td>
<td>Top Drawing 07600002 Revision C dated August 18, 2004 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<tr>
<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
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<td>Associated Technical Documentation</td>
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<td>EASA.IM.R.S.01070</td>
<td>DEVORE AVIATION CORPORATION OF AMERICA</td>
<td>FAA STC</td>
<td>SH58735W</td>
<td>AS 365N1/N2, EC 156B/1</td>
<td>Installation of Tail Floodlights and Associated Technical Documentation</td>
<td>Devore Drawing List 3650001 Revision J dated March 16 1999 DCN F-1 dated February 8 1999 &amp; DCN L-1 dated October 22 2001 or later approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
</tr>
<tr>
<td>EASA.IM.R.S.01071</td>
<td>DEVORE AVIATION CORPORATION OF AMERICA</td>
<td>FAA STC</td>
<td>SH68683W</td>
<td>AS 369N1/N2, EC 156B/1</td>
<td>Installation of Forward Facing Recognition Lights</td>
<td>Devore Drawing List 3650003 Revision C dated July 2 2001 or later approved revision. Devore Aviation Corporation RFM Supplement dated June 10 2002 or later approved revision.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
</tr>
<tr>
<td>EASA.IM.R.S.01072</td>
<td>DEVORE AVIATION CORPORATION OF AMERICA</td>
<td>FAA STC</td>
<td>SH75565W</td>
<td>AS 369N1/N2/3, EC 156B/B1</td>
<td>Installation of a rotor disc light and associated DeVore Aviation Corporation Top Drawing Rotor Disc Light Inspection; AerospatialeDauphin 3650003 revision E dated August 21 2001 or later approved revision. DeVore Aviation Corporation RFM Supplement dated June 10 2002 or later approved rev...</td>
<td>Devore Aviation Corporation Top Drawing- Rotor Disc Light InspectionAerospatialeDauphin 3650003 revision E dated August 21 2001 or later approved revision/Devore Aviation Corporation RFM Supplement dated June 10 2002 or later approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
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<tr>
<td>EASA.IM.R.S.01074</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC</td>
<td>SH84-30, ISSUE 2</td>
<td>ROBINSON R22 BETA</td>
<td>Validation of TCCA STC SH84-30 for the installation of Cabin Floor Protectors.</td>
<td>Installation Instructions IN- D022-720, Revision A dated 6.04.2004 or later TCCA approved revisions/Instructions for Continued Airworthiness ICA-D022-720 Revision 0 or later TCCA accepted revisions.</td>
<td>1. Installation and Maintenance in accordance with the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall...</td>
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<tr>
<td>EASA.IM.R.S.01075</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC</td>
<td>SH84-40, ISSUE 1</td>
<td>ROBINSON R44 II</td>
<td>Validation of TCCA STC SH84-40 for the installation of Cabin Floor Protectors.</td>
<td>Installation Instructions IN- D044-719, Revision A dated 4.06.2004 or later TCCA approved revisions/Instructions for Continued Airworthiness ICA-D044-719 Revision 0 or later TCCA accepted revisions.</td>
<td>1. Installation must be performed in accordance with Installation Instructions IN- D044-719.2. Maintenance in accordance with Instructions for Continued Airworthiness ICA- D044-719.3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall...</td>
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<tr>
<td>EASA.IM.R.S.01076</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC</td>
<td>SH85-7, ISSUE 1</td>
<td>ROBINSON R44 II</td>
<td>Validation of TCCA STC SH85-7 for the installation of a Cold Weather Baffle Kit for Cold Weather operation of Robinson R44 and R44II.</td>
<td>Installation Instructions IN- D044-717 Rev. B TCCA Approved 25. 01.2005 or later TCCA approved revisions. Instructions for Continued Airworthiness ICA-D044-717 Rev. 0 TCCA accepted 01.02.2005 or later TCCA accepted revisions.</td>
<td>1. Installation must be performed in accordance with Installation Instructions IN- D044-717.2. Maintenance in accordance with Instructions for Continued Airworthiness ICA- D044-717.3. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall...</td>
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<tr>
<td>EASA.IM.R.S.01078</td>
<td>HELLYN</td>
<td>TCCA STC</td>
<td>SH58-4, ISSUE 1</td>
<td>EUROCOPTER FRANCE AS 359F2/ EUROCOPTER FRANCE AS 359F1/ EUROCOPTER FRANCE AS 359F/ EUROCOPTER FRANCE AS 359E</td>
<td>Installation of Anticipator of Governor CaltechAvixs Refer to installation manual IM, PA 04 015 91 01 issue 1 or later approved</td>
<td>Refer to installation manual IM, PA 04 015 91 01 issue 1 or later approved</td>
<td>1. Refer to instructions for continued airworthiness ICA/PA 04 015 05 01 issue 1 or later approved and associated maintenance instructions.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility...</td>
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<td>EASA STC no.</td>
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<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
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<td>Associated Technical Documentation</td>
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<td>EASA.IM.R.S.01082</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>SR00688NY</td>
<td>SIKORSKY S-76C SIKORSKY S-76B SIKORSKY S-76A</td>
<td>Installation of Medical Interior</td>
<td>Keystone Helicopter Installation Drawing List IDL00136 Revision C dated March 15, 1999 or later approved revision Keystone Helicopter RPM Supplement FM0136, FM00136.2 &amp; FM00136.4 dated September 26, 1997 or later approved revision for Sikorsky S-76 A, B &amp; C Models is required for this installation on</td>
<td>12/05/2006</td>
<td>Active</td>
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<tr>
<td>EASA.IM.R.S.01083</td>
<td>METRO AVIATION, Inc.</td>
<td>FAA STC SH3880SW</td>
<td>EUROCOPTER DEUTSCHLAND MBB-BK</td>
<td>Installation of a Vapor Cycle Air Conditioner in accordance with FAA STC SH3880SW</td>
<td>FAA approved Flight Manual Supplement EC145 AC FMS</td>
<td>31/05/2005</td>
<td>Active</td>
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<tr>
<td>EASA.IM.R.S.01084</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SR00597RC</td>
<td>206A, 206B</td>
<td>STC ST00597RC Increase in takeoff weight from 1,451 Kg (3,200 lbs) to 1,520 Kg (3,350 lbs) when modified in accordance with Bell Model 206 Series Service Instructions BHT-206-S1-129 dated July 12, 2004, or later FAA approved revision</td>
<td>Rotorcraft Flight Manual Supplement for Increased Internal Gross Weight to 3350 lbs., BHT-206B-FFS-37, Bell Helicopter Textron Inc., Original Revision, dated July 12, 2004 Instructions for Continued Airworthiness (ICA) for Bell Model 206B Series helicopters are incorporated in Bell Helicopter Textron</td>
<td>01/02/2006</td>
<td>Active</td>
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<tr>
<td>EASA.IM.R.S.01085</td>
<td>DART AEROSPACE Ltd.</td>
<td>SH05-10, ISSUE 1</td>
<td>BELL 412 EP BELL 412 CT BELL 412 BELL 212B-1 BELL 212 BELL 206B</td>
<td>Spacecoor Installation</td>
<td>Installation must be in accordance with Dart Aerospace Installation Instruction SN-D412-694 rev A or later approved revision Maintenance must be in accordance with Dart Aerospace Instruction for Continued Airworthiness ICA-D412-694 Rev A TCCA accepted or later approved revision</td>
<td>02/01/2006</td>
<td>Active</td>
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<td>EASA.IM.R.S.01086</td>
<td>DART AEROSPACE Ltd.</td>
<td>SH04-3B, ISSUE 2</td>
<td>SIKORSKY S-76C SIKORSKY S-76A</td>
<td>Installation of Cargo Mirrors</td>
<td>Dart Aerospace Ltd. Drawing No IN-0076-612 Revision A dated June 26, 2002 or later approved revision</td>
<td>25/01/2006</td>
<td>Active</td>
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<td>EASA STC No.</td>
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<td>TC╱EASA No.</td>
<td>Certification basis</td>
<td>System Status</td>
<td>Installation/Operation/Design Classification</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.R.S.01087</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH05-14</td>
<td>Installation of Replacement Window Kit in accordance with TCCA STC SH05-14.</td>
<td>Active</td>
<td>Installation Instructions SH05-14 D022-714, Revision A approved 26.03.2004 or later approved revisions.Instructions for Continued Airworthiness ICA-D022-714 Revision 0 Transport Canada accepted 18.03.2005 or later. Transport Canada accepted revisions.Flight Manual Supplement FMS-D022-714 Rev A Transport.</td>
<td>1. Installation, Operation and Maintenance must be held in accordance with the above referenced Technical Documentation. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>14/06/2005</td>
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<tr>
<td>EASA.IM.R.S.01088</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH05-13</td>
<td>Installation of Replacement Window Kit in accordance with TCCA STC SH05-13.</td>
<td>Active</td>
<td>TCCA approved Flight Manual Supplement FMS-D044-713.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>13/06/2005</td>
</tr>
<tr>
<td>EASA.IM.R.S.01089</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SR0928RC-D</td>
<td>American Eurocopter LLC Modification FAA STC SR0928RC-D Installation of a Garmin Global Positioning System (GPS) - GPS500.</td>
<td>Active</td>
<td>AE Drawing List 155A82-807 Revision E dated April 24 2003 or later approved revision. RFM Supplement 155F/MS.003 dated April 29 2003 or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>15/06/2005</td>
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<td>EASA.IM.R.S.01090</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SR0929RC-D</td>
<td>Installation of horizontal tip position/antiknock index (RON+MON)/2 per ASTM D4814 as noted on placard to be installed in accordance with the FAA STC SR0929RC-D dated 22/9/2003. AIA Master Drawing List Report AA-01093, Revision B dated 19/3/2003.</td>
<td>Active</td>
<td>AE Drawing List 155A82-807 Revision B dated April 22, 2004 or later FAA approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>20/07/2005</td>
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<td>EASA.IM.R.S.01091</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA STC SR0932RC-D</td>
<td>Installation of Passenger photo Window for Bell 407 Model Helicopters in accordance with the FAA STC SR0932RC-D dated 22/9/2003.AIA Master Drawing List Report AA-01095, Revision B, dated 19/3/2003.</td>
<td>Active</td>
<td>Instructions for continued airworthiness AA-01093 revision B dated 5/12/2001. Installation Instructions Passenger Window Kit AA-01091 dated 18/9/2003, Drawing list AA-1095 dated 3/7/2001.</td>
<td>1. The compatibility of this design change with other previously approved modifications must be determined by the installer.2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>22/08/2005</td>
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<td>EASA.IM.R.S.01097</td>
<td>AGUSTAWESTLAND PHILADELPHIAICO ROPORATION</td>
<td>TCCA STC SH05HE</td>
<td>Installation of Alternate Static Pressure Source</td>
<td>Active</td>
<td>Configuration A: In accordance with Agusta drawings 109-A720-14 and 109-A073-65-77 all dated 27-7-84.Configuration B: Agusta kit No. 109-A720-14 in accordance with Agusta Service Instruction No. 109-A-005 dated 11 February 1985.</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer. Configuration A is eligible only on aircraft S/N 7268. When altered...</td>
<td>04/07/2005</td>
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## List of EASA Supplemental Type Certificates

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<td><strong>EASA.IM.R.S.01099</strong></td>
<td>HERITAGE</td>
<td>FAA STC</td>
<td>SR00550NY</td>
<td>EUROCOPTER</td>
<td>Installation of a Cocoon wall panel system.</td>
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<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>18/07/2005</td>
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<td><strong>EASA.IM.R.S.01100</strong></td>
<td>AIRBUS</td>
<td>FAA STC</td>
<td>SR09282RC-D</td>
<td>155 B1</td>
<td>Installation of a Honeywell Enhanced Ground Proximity Warning System.</td>
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<td>11/08/2005</td>
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<td><strong>EASA.IM.R.S.01102</strong></td>
<td>BELL</td>
<td>FAA 407</td>
<td>SR09239RC</td>
<td>407</td>
<td>Installation of a Steerable Landing Light.</td>
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<td>10/05/2007</td>
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<td><strong>EASA.IM.R.S.01108</strong></td>
<td>SIKORSKY</td>
<td>FAA STC</td>
<td>SR01959NY-D</td>
<td>SIKORSKY S-76C</td>
<td>Installation of Rockwell Collins TDR-94D Transponder with Elementary Surveillance</td>
<td>1. Scotty Microwave TDR-94D. Transponder with Elementary Surveillance with Enhanced Airborne Surveillance System. 2. Bell Helicopter Textron, Inc. High Skid Landing Gear 412-050-015-111/-112 or later approved revision.</td>
<td>Installation Drawing List ID000056 rev A dated May 11, 2005 or later approved revision. Flight Manual Supplement FM00562 rev dated May 20, 2005 or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>13/03/2006</td>
<td>Active</td>
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**Notes:**
- FAA STC: FAA Supplemental Type Certificate
- EASA STC no.: EASA Supplemental Type Certificate Number
- Certification basis: Basis for the certification of the STC
- Associated Technical Documentation: Details of the technical documentation associated with the STC
- Limitations/Conditions: Specific limitations or conditions associated with the STC
- Date of Issue: Date the STC was issued
- Status: Current status of the STC
- System Record: Unique identifier for the STC in the EASA database

**Related Links:**
- EASA.europa.eu

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**Image:**
- None
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<td>EASA.IM.R.S.01109</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SRO105NY-D</td>
<td>SIKORSKY S-76C</td>
<td>Installation of baggage access panel (ground use only), forward facing seat, seat pan and wet floor.</td>
<td>Installation Drawing List IDL00921 rev. ** dated May 16, 2005 or later approved revision/Flight Manual Supplement FM00921 rev - dated May 24, 2005 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>KEYSTONE HELICOPTER SERVICES</td>
<td>FAA STC SRO195NY-D</td>
<td>S76A/B/C</td>
<td>Installation of Grimes Search Lights</td>
<td>Installation Drawing List IDL00928 rev. A dated May 16, 2005 or later approved revision/Flight Manual Supplement FM00928 rev - dated May 24, 2005 or later approved revision.</td>
<td>This SIC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>EASA.IM.R.S.01111</td>
<td>KEYSTONE HELICOPTER SERVICES</td>
<td>FAA STC SRO195NY-D</td>
<td>S76C</td>
<td>Installation of a Vertical Visibility Door Window in the pilot's door</td>
<td>Installation Drawing List IDL00929 rev. A dated May 4, 2005 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise suspended or revoked.</td>
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<td>EASA.IM.R.S.01112</td>
<td>KEYSTONE HELICOPTER SERVICES</td>
<td>FAA STC SRO1962NY-D</td>
<td>S76B/C</td>
<td>Installation of Rockwell Collins MDP-124F (V2) Direction Finder System</td>
<td>Installation Drawing List IDL00895 rev. ** dated May 16, 2005 or later approved revision/Flight Manual Supplement FM00895 rev - dated May 24, 2005 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise suspended or revoked.</td>
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<td>EASA.IM.R.S.01113</td>
<td>KEYSTONE HELICOPTER SERVICES</td>
<td>FAA STC SRO1963NY-D</td>
<td>S76B/C</td>
<td>Installation of the WesCam i2DS200 Camera System</td>
<td>Installation Drawing List IDL00896 rev. ** dated May 16, 2005 or later approved revision/Flight Manual Supplement FM00896 rev - dated May 24, 2005 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise suspended or revoked.</td>
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<td>EASA.IM.R.S.01114</td>
<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>FAA STC SRO0851LA</td>
<td>AS 355 E/F/F1/F2/N AS 350 D1/D1B1B2/B3/B4.</td>
<td>Installation of Apical Emergency Float kit, Part Number 20416, in accordance with the following FAA approved Apical Industries, Inc. Master Document List and Rotorcraft Flight Manual Supplement. - For AS 350 series : MDL n° MG356(4) rev. 1, dated December 1st, 2004 or later FAA approved revision...</td>
<td>- Apical AS 350 RFMS n° FMS-356(4), rev. B dated 15/11/05 inalter EASA approved revision/-Apical AS 355 RFMS n° FMS-355 (2) rev. B dated 15/11/05 or later EASA approved revision/ Apical Installation and Packing Instructions n° 8355-100 rev. E, dated October 26th, 2004, or later FAA approved revision...</td>
<td>- This approval requires the inspection and overhaul schedules to be performed as mandated by the Apical Instructions for Continued Airworthiness n° ICA 355-1, rev. G, dated December 3rd, 2004 or later FAA approved revision. 2. This installation should not be incorporated in any aircraft unless it is...</td>
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<td>APLICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
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<td>FAA STC SRO1687LA</td>
<td>EC 130 B4</td>
<td>Installation of Apron Emergency Float kit p/n 604-0018-100 or Emergency Float kit with Lifeboat p/n 604-0018-300.</td>
<td>Definition from Apron Master Document List n° MDI31-1, revision, dated 04/28/05, or later FAA approved revisions.Installation Aprical Installation Instructions for EC 130 B4 Emergency Float kit n° 11130-1, revision D, dated 12/20/04, or later revisions/Apical Installation Instructions for EC 130 B4 Emer...</td>
<td>This STC is the EASA variation of FAA STC n° SRO1687LA, dated March 22nd, 2005.</td>
<td>29/04/2008</td>
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<td>EASA.IM.R.S.01116</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
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<td>FAA STC SRO2074AT</td>
<td>427</td>
<td>Manufacture of Automatic Door Opener for Bell 427 Model Helicopters in accordance with the FAA STC SRO2074AT, All Master Drawing List Report AA-98109 dated 30/10/1998. FAA STC SR 02547AT dated 14/02/2000 (Atlanta Aircraft Certification Office).</td>
<td>- PMA No. PQ0150CE dated 15/02/2000- Instructions for continues airworthiness AA-9904 dated 05/12/2001- Draw door opener kit AA-98099 dated 29/04/1999- Passenger door opener kit AA-98100 dated 12/12/2002- Baggage door opener kit AA-99105 dated 24/06/2002</td>
<td>1. The compatibility of this design change with other previously approved modifications must be determined by the installer. 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft...</td>
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<td>EASA.IM.R.S.01117</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
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<td>FAA STC SHB045W-D</td>
<td>EC 155 B1</td>
<td>Installation of a Universal Navigation UNIS-1F Flight Management System for IFR enroute &amp; non-precision approaches.</td>
<td>AEC Drawing List 36057-9063 Revision B dated September 13 1996 or later approved revision.RFM Flight Manual Supplement/RFS 6919-026 (dated November 15 1996 or later approved revision).</td>
<td>American Eurocopter Drawing List 15856.8088 revision A dated May 30 2005 or later approved revision.Rotorcraft Flight Manual Supplement dated July 8 2005 or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/...</td>
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<td>EASA.IM.R.S.01122</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
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<td>FAA STC SH1260NW</td>
<td>EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B</td>
<td>Electronic Model E-69 Cargo Hook Load Cell System p/n 200-295-00, in accordance with Onboard Systems Master Drawing List n° &quot;055-019-00, revision 10 dated 15/06/04 approved by FAA or later FAA approved revision.</td>
<td>Installation Instructions : Onboard Systems Owners Manual n° 125-114-00, revision 0 dated 15/06/04 approved by FAA or later FAA approved revision.</td>
<td>1. - The present STC is the validation of the FAA STC SH126099 last amended on 15/03/05, limited to Eurocopter AS 350 B, B1, B2 and D models. 2. - The system covered by this STC can be installed on AS 350 helicopters equipped with Eurocopter sling systems, or on Eurocopter AS 350 B2 or earlier swin...</td>
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<td>EASA.IM.R.S.01123</td>
<td>AERO DESIGN LTD</td>
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<td>TCCA STC SH02.2L, ISSUE 2</td>
<td>AS 350 B1 AS 350 E</td>
<td>Installation of DynaNav Sailsbag System.</td>
<td>Definition and installation AEROS Design Ltd. Document Control List OCL444, Revision 2, or later Transport Canada (TCCA)approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>AIR COMM CORPORATION LLC</td>
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<td>SR00491DE</td>
<td>EC 120 B</td>
<td>Installation of Calon Air Conditioning system. The present STC is the validation of the FAA STC n° SR00491DE.</td>
<td>Definition: Air Comm Corporation Master Drawing List report DL-EC120, revision D, dated December 26, 2002 approved by FAA or later approved revisions. Installation: Air Comm Corporation Master Drawing List report n° DL-EC120, revision D, dated December 26, 2002 approved by FAA or later approved revisions.</td>
<td>24/12/2006</td>
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<td>EASA.IM.R.S.01132</td>
<td>AIR COMM CORPORATION LLC</td>
<td></td>
<td>SR00483DE</td>
<td>A119</td>
<td>Air Conditioning System for Agusta A 119FAA STC SR00483DE issued on 11th January 2002 and amended on 24th January 2006</td>
<td>This approval should not be extended to rotorcraft of this model on which other previously approved modifications are incorporated unless it is determined that the relationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person...</td>
<td>02/03/2006</td>
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<td>EASA.IM.R.S.01136</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>FAA STC SH4908NM</td>
<td>MD HELICOPTERS 36E MD HELICOPTERS 369D</td>
<td>Installation of Model E-51 Cargo Hook Load Cell System PN 200-034-00. This STC is the validation of the FAA STC SH4908NM amended on 05/11/1990 and limited to MDHI 369 D and E models</td>
<td>Master Drawing List (MDL) Onboard Systems MDL No. 155-014-00 Revision 1 dated 10/10/1990 approved by FAA or later FAA approved revision Onboard Systems Owners Manual No. 120-017-00 Revision 1 dated 10/10/1990 approved by FAA or later FAA approved revision Installation Instructions Onboard Systems Owners Manual No. 120-017-00 Revision 1 dated 10/10/1990 approved by FAA or later FAA approved revision</td>
<td>1. This STC is to be installed on aircraft models specified in “applicability” section above, equipped with MDHI Cargo Hook Kit No.369H90072, using MDHI cargo hook manual release PN 14323.2 only. Installation, Inspection and Operation must be done in accordance with the documents specified in thee...</td>
<td>12/10/2005</td>
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<td>EASA.IM.R.S.01137</td>
<td>SKYTRAC SYSTEMS LTD.</td>
<td>TCCA STC SH05-42</td>
<td>SEE APPROVED MODEL LIST</td>
<td>SkyTrac ISAT-100 Airborne Position/Data Communicator System Installation.TCCA STC SH05-42 Issue no 1 dated 12 September 2005</td>
<td>SkyTrac Systems Inc. Master Document List 250-55997-00 Rev. 01, dated April 13, 2001, or later approved revision Supplement Instructions for Continued Airworthiness are provided in SkyTrac Document 270-55997-00 Rev. 00, dated September 1, 2005 or later approved revisions.</td>
<td>1. Equipment configuration: ISAT-100 (no mod.) or later TCCA approved major mod.2. This STC references specified interfacing equipment that is approved via the data. The ISAT-100 equipment is capable of being interfaced to traditional aircraft systems for which approval is beyond the...</td>
<td>13/02/2006</td>
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<td>EASA.IM.R.S.01138</td>
<td>CARSON HELICOPTERS, INC.</td>
<td>FAA STC SH639NE</td>
<td>S61 N</td>
<td>Installation of 11,000 pound capacity Cargo Sling in accordance with Carson Services, Inc., Drawing List Report No S61N-1A Revision 0 dated January 10, 2005</td>
<td>Carson Services, Inc. FAA approved Flight Manual Supplement No. 1, as approved February 2, 2005 (or later approved revision) to FAA approved Supplement Flight Manual Supplement No. 15-Carson Services, Inc. FAA approved Flight Manual Supplement No. 1, as approved February 2, 2005 (or later approved revision)</td>
<td>None</td>
<td>06/07/2009</td>
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<td>AS 350 B, AS 350 B1, AS 350 B2, AS 350 B3, AS 350 BA, AS 350 D</td>
<td>EASA IM.R.S.01150</td>
<td>EUROCOPTER DEUTSCHLAND BO-105, Ambulance Rescue System The FOCA STC Z 25-20-49 (ambulance rescue system) was validated by the LBA under the number 0566/3025 dated 24 February 1995, approving installation onto the BO105The STC is re-issued as EASA STC to extend applicability to BO105S.</td>
<td>Flight Manual Supplement FMS-10-66 This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/ engine configurations shall be determined by the installer.</td>
<td>11/11/2005</td>
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<td>Pratt &amp; Whitney</td>
<td>SCHWEIZER AIRCRAFT 2690</td>
<td>EASA IM.R.S.01163</td>
<td>Installation of Pratt &amp; Whitney Engine Services, Inc. Intelligent Electronic Control System and Gimble Ensemble Monitor PN DPL-K-010-14 Pratt &amp; Whitney Engine Services, Inc. Intelligent Electronic Control System and Gimble Ensemble Monitor PN DPL-K-010-14</td>
<td>Pratt &amp; Whitney Engine Services, Inc. Intelligent Electronic Control System and Gimble Ensemble Monitor PN DPL-K-010-14 Pratt &amp; Whitney Engine Services, Inc. Intelligent Electronic Control System and Gimble Ensemble Monitor PN DPL-K-010-14 1. Engine must be modified in accordance with STC SE0027EN, which must be EASA approved or accepted prior to installation. 2. Reference STC SE0027EN for any additional limitationsapplicable to the engine installation3. This system is designed as an electronic aid for engine “hot start” prevention a...</td>
<td>14/03/2008</td>
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<td>EASA.IM.R.S.01166</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
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<td>FAA STC SR00368SE</td>
<td>BELL 407 BELL 206-4 BELL 206-3 BELL 206-1 BELL 206L</td>
<td>Cargo hook installation n. 200-215-00</td>
<td>- FAA approved Onboard Systems Master Drawing Lot n. 155-029-00 dated June 26, 2000 or later FAA approved revision. - FAA approved Onboard Systems Owner’s manual n. 120-046-00 dated June 28, 2000, or later FAA approved revision. - FAA approved Onboard Systems Service Manual n. 122-001-00 dated Jun...</td>
<td>Approval of this change in type design applies to only those Bell model rotorcraft listed above, which were previously equipped with an FAA approved installation of Bell cargo hook suspension systems 206-706-341-5, 206-706-341-101, 206-706-341-109 and either Breeze-Eastern cargo hooks 17145-2 or 17...</td>
<td>27/09/2006</td>
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<td>EASA.IM.R.S.01167</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
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<td>FAA STC SR00826SE</td>
<td>MD HELICOPTERS 500N MD HELICOPTERS 369HS MD HELICOPTERS 369HM MD HELICOPTERS 369HE MD HELICOPTERS 369FF MD HELICOPTERS 369F MD HELICOPTERS 369D MD HELICOPTERS 369D</td>
<td>Installation of Onboard Systems Model 260-264-00 Cargo Hook.sEASA validation of FAA STC n2 SR00826SE, amended January 13th, 2003 or later approved revision. This STC is classified ‘Basic’ by the FAA and is approved under the post type Validation Principles (PTVP).</td>
<td>Onboard Systems Master Drawing Lot n0155-061-00, revision 7 dated June 18th, 2006 or later approved revision. Onboard Systems Owner's Manual n0120-096-00, revision revised September 22nd, 2005 or later approved revision. Onboard Systems Instruction for Continued Airworthiness n0123-003-00, revision 2...</td>
<td>This cargo hook is not approved for human external load. Approval of this installation applies to the above listed model/rotorcraft only, which were previously equipped with an approved installation of McDonnell Douglas cargo hook suspension systems and cargo hooks shown in the table below, or Onboa...</td>
<td>08/07/2008</td>
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<td>EASA.IM.R.S.01168</td>
<td>LUFTTRANSPORT AS</td>
<td></td>
<td>- EUROCOPTER FRANCE AS 365N SERI</td>
<td>Mast Measurement System / System for Measurement of ground stations fixed radio antenna for correct frequencies as signal effects</td>
<td></td>
<td></td>
<td>None</td>
<td>None</td>
<td>12/06/2006</td>
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<td>EASA.IM.R.S.01169</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
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<td>FAA STC SR00575SE</td>
<td>ROBINSON R22 MARINER ROBINSON R22 BETA ROBINSON R22 ALPHA ROBINSON R22</td>
<td>Cargo Hook Suspension System</td>
<td>FAA approved Onboard Rotorcraft Flight Manual Supplement No121-022-00, dated September 29, 2005 or later approved revision/FAA approved Onboard Systems Owner's Manual No121-070-00, dated October 11, 2004 or later approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>02/01/2006</td>
<td>Active</td>
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<tr>
<td>EASA.IM.R.S.01171</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>REV. 1</td>
<td>FAA STC SR01044SE</td>
<td>ROBINSON R44 II ROBINSON R44</td>
<td>Cargo hook suspension system i.a.w. US STC SR01044SE Note:Revision 1 of this STC is made for correction of FAA original STC reference. This supersedes previous issuance of the STC.</td>
<td>- FAA approved Onboard Systems Rotorcraft Flight Manual Supplement No. 121-047-00, dated April 22, 2005 or later approved revision for the Robinson R44 rotorcraft. - FAA approved Onboard Systems Rotorcraft Flight Manual Supplement No. 121-017-00, dated April 22, 2005 or later approved revision...</td>
<td>29/04/2009</td>
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### List of EASA Supplemental Type Certificates

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<tr>
<td>EASA.IM.R.S.01179</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH05-5 ISSUE 2</td>
<td>BELL 407, BELL 206L-4, BELL 206L-3, BELL 206L-1, BELL 206L, BELL 206A</td>
<td>Landing Gear Crossribies</td>
<td>ACP-D206-667 rev AACP- D407-667 rev AAACR- D206-667 rev A</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;, Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
<td>None</td>
<td>03/07/2006</td>
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<tr>
<td>EASA.IM.R.S.01180</td>
<td>DART AEROSPACE Ltd.</td>
<td>SH04-54 ISSUE1</td>
<td>BELL 412, BELL 214B-1, BELL 214B, BELL 212, BELL 205A, BELL 205A, BELL 204B</td>
<td>Installation of Fuel Purge Canister</td>
<td>Installation Instructions: IMA-D412-706 Revision A, dated November 30, 2004 or later approved revision.Instructions for Continued Airworthiness: ICA-D412-706, dated November 30, 2004 or later approved revision</td>
<td>None</td>
<td>None</td>
<td>26/04/2007</td>
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<td>1018185</td>
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<tr>
<td>EASA.IM.R.S.01185</td>
<td>AEROLITE AG</td>
<td>-</td>
<td>EC135</td>
<td>&quot;EMS Equipment P/N 135138-502This major modification coversan EMS equipment inoluding medical seats, stretcher, oxygen/linnder rack, storage cabinet and Oxygen and Propas retainsome components, identified in the equipement list EUL-05-033 and already approved under Swedish Civil Aviation Admin...</td>
<td>Flight manual supplement FMS-06-003</td>
<td>1. Limited to S/N 1052 No Smoking when Medical Oxygen equipment is installed. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/engine configurations shall be determined by the...</td>
<td>22/03/2006</td>
<td>Active</td>
<td>1018183</td>
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**Notes:**
- STC: Supplemental Type Certificate
- EASA: European Aviation Safety Agency
- FAA: Federal Aviation Administration
- TCCA: Transport Canada Certification Authority
- ICA: Industry Canada Air
- AIC-PA-04-021-91-01: Installation Instruction for the Bell 407 STC
- IM-PA-04-021-95-01: Instruction for Continued Airworthiness for the Bell 407 STC
List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<td>EASA.IM.R.S.01186</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>10016827</td>
<td>10016822</td>
<td>S-76C</td>
<td>Installation of Bristow Helicopter’s Ltd Automatic floatingDeployment System (AFDS) in accordance with Keystone Installation drawing list IDL00601 Rev. dated 11/27/02</td>
<td>Keystone Bristow Helicopter Corp. FAA approved rotorcraft flight manual Supplement No. FM00601, no revision, FAA approved December 15, 2002. is required with this modification</td>
<td>This STC is valid only for STC SIKORSKY helicopter model from S/N 750011 and subsequent</td>
<td>20/02/2006</td>
<td>Active</td>
<td>1011823</td>
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<tr>
<td>EASA.IM.R.S.01188</td>
<td>METRO AVIATION, Inc.</td>
<td>REV. 1</td>
<td>FAA STC SR09163RC</td>
<td>EUROCOPTER EC 135</td>
<td>Installation of mechanically driven vapor cycle air condition system i.a.e. FAA STC SR09163RC as amended by 18-06-2007 Major change to EASA STC EASA.IM.R.S.01188; addition of models and incorporation of technical modifications.</td>
<td>Flight Manual Supplement 135AC-7, Revision B, dated 15 May 2006 or later approved revision or. Flight Manual Supplement 135AC-8, Revision A, dated 16 June 2007 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>14/04/2008</td>
<td>Active</td>
<td>1011823</td>
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<tr>
<td>EASA.IM.R.S.01196</td>
<td>PARAVION TECHNOLOGY, Inc.</td>
<td>FAA STC SR00338DE</td>
<td>BELL 407</td>
<td>Installation of Anti-Torque Extention Adjustment System</td>
<td>Master Drawing List DL-407/PAK-100 rev B</td>
<td>None</td>
<td>1. Required maintenance, as prescribed in the applicable airplane or engine Instructions for Continued Airworthiness, shall not be predicated on the data obtained from the Pratt &amp; Whitney Engine Services, Inc. monitoring/recording system. 2. This system shall not be used to supersede or replace the pi...</td>
<td>03/07/2008</td>
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<td>EASA STC no.</td>
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<td>EASA.IM.R.S.01201</td>
<td>STARFLEX LLC.</td>
<td>FAA STD-SH7962SW-D</td>
<td>EUROCOPTER FRANCE AS 359F2</td>
<td>Installation of two Allison 250-C20R engines and enhanced anti-torque system in accordance with AHC drawing list 35SAE2-175S Revision D dated November 5, 1990 or later FAA approved revision</td>
<td>Instructions for Continued Airworthiness: Starflex LLC document reference C20RF2MM, edition 1 amendment 0 dated 11 January 2007, or later revision</td>
<td>1. The present STC is the validation of the FAA STC SH7962SW-D, last amended on 01 February 2006.2. Compatibility of this modification with previously approved modifications of the aircraft must be determined by the installer.</td>
<td>16/01/2007</td>
<td>Active</td>
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<td>EASA.IM.R.S.01202</td>
<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>FAA STC-S3R01456LA</td>
<td>412, 412EP</td>
<td>Installation of Emergency Mid-Float with 10-man Liferaft Kit; Installation of Apical EMERGENCY Mid-Float with 10-man Liferaft Kit</td>
<td>Master Document List MD-412-31, Rev. K, dated 3 February 2006 or after approved revision Instruction for continued Airworthiness ICA-412-31 Rev E, dated 13 April 2005 or later approved revision</td>
<td>Bell Helicopter Textron, Inc. Emergency Float Kit Part Number 412-706-004-137, or 412-706-004-149 or Apical Industries, Inc. Emergency Float Kit Part Number 634.9601 per FAA STCSR01644LA, when approved by EASA must be installed prior to the installation of this STC. This approval is not valid for military aircraft.</td>
<td>04/09/2006</td>
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<td>EASA.IM.R.S.01203</td>
<td>STARFLEX LLC.</td>
<td>FAA STC-SH7961SW-D</td>
<td>AS 355 F1</td>
<td>Installation of two Allison 250-C20R engines in accordance with AHC drawing list 35SAE2-175S Revision H dated November 5, 1990 or later FAA approved revision</td>
<td>Instructions for Continued Airworthiness: Starflex LLC document reference C20RF1MM, edition 1 amendment 0 dated 11 January 2007, or later revision Operation: AHC Rotorcraft Flight Manual Supplement reference 35S-FNS-011, dated 01 February 1991 approved by FAA, or later FAA approved revision</td>
<td>1. The present STC is the validation of the FAA STC SH7961SW-D, last amended on 01 February 2006.2. Compatibility of this modification with previously approved modifications of the aircraft must be determined by the installer.</td>
<td>16/01/2007</td>
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<td>EASA.IM.R.S.01208</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC-SH7014SW-D</td>
<td>SA 365 N1 AS 365 N2, AS 365 N3</td>
<td>Installation of vertical fin anti-collision light, in accordance with AEC Drawing List No. 365A73-9027, revision E dated 09 June 2003, or later FAA approved revision</td>
<td>--</td>
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<td>EASA.IM.R.S.01209</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC-SH5495SW-D</td>
<td>SA 365 N1 AS 365 N2, AS 365 N3</td>
<td>Installation of supplement stove light, in accordance with AEC Drawing List No. 365A73-9027, revision L dated 06 June 2003, or later FAA approved revision</td>
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<td>AIRBUS HELICOPTERS, Inc.</td>
<td>EASA.IM.R.S.01210</td>
<td>SH37003W-D</td>
<td>AS 355 N2, SA 365 N1</td>
<td>Installation of a Bendix/King GPS KLN 90B or KLN 900 for IFR en route operations in accordance with AEC Drawing List 365A73-9060, revision C, dated 02/03/2005, or later EASA approved revision.</td>
<td>AEC RFM Supplement No. 365FMS-047, original issue or later EASA approved revision.</td>
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<td>AERO SEATS AND SYSTEMS, INC</td>
<td>EASA.IM.R.S.01214</td>
<td>SR04955RC</td>
<td>S-76A</td>
<td>Installation of External Liferaft Deployment System</td>
<td>CHC Scotia Report Number CHCE-2500-006 mod asi 500-100 01/02 Revision - dated January 18, 2006 or later FAA approved revision; RFM Supplement S-76-500-FMS Revision IR dated March 1, 2006 or later approved revision.</td>
<td>1. HeliPod USA pilot's operating Handbook Supplement No 1, rev A dated 23 March 2005, for Robinson R22 helicopters is required and must be carried during all flights. 2. The approval should not be extended to other aircraft of these models or which other previously approved modifications are incorporated...</td>
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<tr>
<td>AERO SEATS AND SYSTEMS, INC</td>
<td>EASA.IM.R.S.01218</td>
<td>SR09355RC</td>
<td>S-76A, S-76B, S-76C</td>
<td>Installation of Passenger Seats</td>
<td>ASI-6409 - Installation Instructions</td>
<td>None</td>
<td>10/05/2006</td>
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<td>HELL-TRADE CORPORATION</td>
<td>EASA.IM.R.S.01219</td>
<td>SR01592SE</td>
<td>BELL 412, BELL 214ST, BELL 212, BELL 205B, BELL 205A-1</td>
<td>Installation and Maintenance of Goodrich Internal Hoist FAA STC SR01592SE</td>
<td>Master Drawing List H-415-MDL dated August 10, 2005 or later approval RFM Supplement FMS-415-29-1581 dated August 12, 2005 or later approved revision</td>
<td>In order to operate the internal hoist in the referenced airframes, the following provision kits must be installed: HOIST PIN BELL PROVISION KIT BELL SERVICE INSTRUCTION HELICOPTER S/N EFFECTIVITY YTC-42305A-205-706-001-1 BHT-205-SI-44 205A-1 30001-20028HTC-42305A 205-70.</td>
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<td>EASA.IM.R.S.01221</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>TCGA STC SH05-18</td>
<td>AS 350 B/ B1/ B2/ B3/ B4/ D/D1</td>
<td>Installation of forward short high-visibility doors (350-20414)</td>
<td>Installation must be in accordance with Eurocopter Canada Limited Installation Procedure n. IP-ECL-112. Rev. 0, TCCA approved April 1, 2005 or later TCCA approved revisions. Configuration must be in accordance with Eurocopter Canada Limited Master Drawing List N. MDL-04-073, Iss. C, TCCA approved April 1, 2005 or later approved revision. Prior to installation of this modification the installer must determine that the installation will not introduce any adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.R.S.01223</td>
<td>AIR CRUISERS COMPANY</td>
<td>FAA STC SR0154NY</td>
<td>EC 120 B</td>
<td>Installation of Air Cruisers Company Emergency Flotation System, (STC Classified as Basic by the FAA)</td>
<td>EDR 20-1367 rev C dated 19 December 2003 - Installation Data List EDN 2921 rev 8 dated 8 August 2003 - Instructions for Continued Airworthiness.</td>
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<td>PRATT &amp; WHITNEY</td>
<td>REV. 1</td>
<td>AGUSTA A-109 E</td>
<td>Installation of a Pratt &amp; Whitney Engine</td>
<td>Instructions for Continued Airworthiness as listed in AML DTU-1-225-1. Revision, dated March 27, 2006, or later approved revision</td>
<td>1. 109E with PW &amp; W 206C Engine;</td>
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<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>FAA STC SR0856LA</td>
<td>BD105 A/C/S</td>
<td>Installation of Apical Emergency Float Kit, Part Number 20430Description: Installation of Emergency Float Kit in accordance with FAA STC SR0856LA. Apical MDL MD105-2 proves for detailed installations either with liftballs or without. Note: the recommendation for approval is given based on a re...</td>
<td>For installation i.a.w MD105-2 Rev. B/ 8-18-1989 Instructions for continued airworthiness: ICA 105-2, Rev A. Flight Manual Supplement FMS-105(2) Rev. A for installation i.a.w. MDL MD 105-2 Rev. F/ 13-08-2004 Instructions for continued airworthiness: ICA105-501 Rev. NI CFTMS-105-501 Rev. A The installation is not approved for helicopters equipped with high skid landing gear.</td>
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<td>07/03/2008</td>
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<td>EASA.IM.R.S.01226</td>
<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>FAA STC SR0932LA</td>
<td>369HS/HE/DE/FF, 500N</td>
<td>Installation of Apical Emergency Float kit p/n 20491-1.</td>
<td>Definition Apical Master Document List DG050-10, revision A, dated 10/12/05, or later FAA approved revisions Installation Apical Installation Instructions, nt15234-10, revision D, dated 08/13/05, or later revisions Operation Apical Flight Manual supplement n6 FMS-900000, revision C, dated April 10th.</td>
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<td>EASA.IM.R.S.01227</td>
<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>FAA STC SR01420A</td>
<td>AS 332 C/L/L</td>
<td>Installation of Apical Industries, Inc. Emergency float assemblies, P/N 20329-105, 20330-106 and 2037, and inflation cylinders, P/N 20330-2</td>
<td>FAA approved Apical Industries, Inc. Master Drawing List MDL332-1, rev. J dated September 18, 2005, or later FAA approved revision FAA approved Apical Industries Inc. Instructions for Continued Airworthiness ICA332-1A, rev. A dated January 31, 2006 or later FAA approved revision</td>
<td>1 Aerazur’s emergency float assemblies P/N 158565 and 158566, and inflation cylinders P/N 158562 (i.e. parts of original emergency float kit approved on Eurocopter AS332C, L, and L1 models) must be installed prior to applying this STC. The interrelated installation configuration between the...</td>
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<td>13/05/2008</td>
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<td>Limitations/Conditions</td>
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<td>EASA.IM.R.S.01229</td>
<td>TELEDYNE TECHNOLOGIES INCORPORATED</td>
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<td>SR0427LA</td>
<td>MD HELICOPTERS 600N</td>
<td>Aircraft Battery Replacement with Alternative Designs</td>
<td>Master Data List Document No. SS500-02, revision Adated Dec 20, 2002 or later approved revision FAA Approved Pilots Operating Handbook Supplement and Flight Manual Supplement No. MS5050-03 PM dated February 27, 2003 or later approved revision</td>
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<td>07/06/2006</td>
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<td>EASA.IM.R.S.01230</td>
<td>L-3 WESCAM INC.</td>
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<td>SH04-29</td>
<td>EUROCOPTER FRANCE AS 355N</td>
<td>MX-15 camera installation (structural provisions only) Note: The revision extends the applicability of STC EASA.IM.R.S.01230 to AS355N.</td>
<td>Installation/Operating Data • Installation must be done in accordance with DECA Aviation Engineering Modification Summary MS02146, Revision 3, or later approved revisions. • The installation must be maintained in accordance with DECA Aviation Engineering Instructions for Continued Airworthiness...</td>
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<td>EASA.IM.R.S.01231</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
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<td>SR01140NY-D</td>
<td>SIKORSKY S-76C</td>
<td>Installation of a Cyclic Control Stick Guard</td>
<td>Keytech Installation Drawing List IDL00644 Revision - dated April 17, 2003 or later approved revision Keystone Helicopter AFM Supplement 00644 no Revision dated April 25, 2003 or later approved revision</td>
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<td>PRECISE FLIGHT, Inc.</td>
<td>FAA STD-0701SE</td>
<td>ROBINSON R22 &amp; R44</td>
<td>Installation of Precise Flight Posite System in accordance with Precise Flight, Inc. Engineering drawings 590P0001, Rev., dated 21 January, 1999 and 590P0002, Rev., dated 26 January 1999 or later approved revision in Robinson R22 model or R44 model helicopters.</td>
<td>RFMS - Precise Flight Inc. FAA approved Rotorcraft Flight Manual Supplement, dated May 18, 1999 or later approved revisions. Installation Instructions and Instructions for Continued Airworthiness &quot;Installation Instructions No. PPR-300&quot; Revision 9 or later revision.</td>
<td>20/01/2007</td>
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<td>EASA.IM.R.S.01238</td>
<td>ONSBOARD SYSTEMS INTERNATIONAL</td>
<td>FAA STC-01164SE</td>
<td>EUROCOPTER FRANCE AS 350 B3 EUROCOPTER AS 350 B2 EUROCOPTER AS 350 B1 EUROCOPTER FRANCE AS 350 B</td>
<td>Fabrication and Installation of Onboard Systems Model 200-280-01 Cargo Hook Swing Suspension System - FAA STC SR01164SE dated October 29, 2004; Master Drawing List STC SR01164SE Doc. Number 155-086-00 dated September 29, 2004 Revision 10-; Owner's Manual for the Cargo Hook Swing Suspensions system on the Eurocopter AS350 Series System, Part Number 200-280-01, Manual Number 120-104-01, S...</td>
<td>1. Approval of this change in Type Design applies only to those Eurocopter AS350 model rotorcraft listed above. 2. This approval should not be extended to other rotorcraft offthese models on which other previously approved modifications are incorporated unless it is determined by the installer...</td>
<td>26/07/2006</td>
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<td>EASA.IM.R.S.01240</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC-011150NY-D</td>
<td>SIKORSKY S-76C</td>
<td>Installation of a Honeywell Chadwick Model 15175 XVP System</td>
<td>Keystone Helicopter Installation Drawing List IDL00892 Revision A dated November 21, 2005 or later approved revision</td>
<td>26/06/2006</td>
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<td>EASA.IM.R.S.01241</td>
<td>APICAL INDUSTRIES, Inc dba DART AEROSPACE</td>
<td>FAA STC-011950LA</td>
<td>S-76 A, B, C</td>
<td>Installation of Apical Industries Inc. LiRailf KI</td>
<td>MD962-2; revision E dated 19 May, 2008 -- Master Document List(RFMs)-76, revision N/C dated 28 June, 2006 -- Rotorcraft Flight Manual SupplementCAT76-2, revision A dated 12 June, 2006 -- Instructions for Continued Airworthiness</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation&quot; Compatibility with other aircraft/engines configurations shall be determined by the installer. This certificate shall remain...</td>
<td>16/06/2007</td>
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<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH96-120</td>
<td>BELL 412 EP</td>
<td>BELL 412</td>
<td>Installation of a Utility Basket in accordance with Transport Canada STC SH96-120, Issue 3, dated May 16, 2006.</td>
<td>- FMS-D205-541 for Bell 412 Models Revision A dated May 16, 2006 or later approved revision; - FMS-D205-541 for Bell 205/210/212 Models revision E dated May 16, 2006 or later approved revision; - FMS-D214-541 for Bell 214 models Revision A dated May 16, 2006 or later approved revision.</td>
<td>1. VFR Only; 2. Doors to be closed at all times during flight; 3. Maximum Load 136 kg (300 lb) or 100kg (220lb) if middle support is not installed; 4. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;.</td>
<td>29/09/2006</td>
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<td>RSG PRODUCTS, Inc.</td>
<td>FAA STC SH3506SW</td>
<td>EUROCOPTER FRANCE EC 130 B4</td>
<td>AS 350 B</td>
<td>Installation of a Belt Driven Freon Air Conditioning System (FAA STC SH3509SW).</td>
<td>- DL-9 rev A dated 12 Sept 1985 – Drawing List, Single Condenser Blower Models A and D; FMS dated 26 June 1985 or later approved revision; - DL-9 rev B dated 10 Jan 1992 – Drawing List, Dual Condenser Blower Models AS and D; FMS dated 29 June 1985 or later approved revision.</td>
<td>1. VFR Operations only; 2. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft engine configurations shall be determined by the install...</td>
<td>26/04/2007</td>
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<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>FAA STC SR0664SE</td>
<td>BELL 206B</td>
<td>Cargo Hook Suspension Kit for Bell 206A and 206B</td>
<td>Cargo Hook Suspension Kit for Bell 206A and 206B</td>
<td>200-233-00 or 200-235-00 (without load weight); 200-234-00 or 200-236-00 (with load weight); MDL 155-041-00 rev 1 dated November 20, 1988 or later approved revision. Owners Manual 120-076-00 revision 1 dated November 20, 1988 or later approved revision. Flight Manual Supplement 120-076-00 dated December...</td>
<td>None</td>
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<td>AIR METHODS CORPORATION</td>
<td>FAA STC SR0003DE-D</td>
<td>AB412 / 412</td>
<td>Installation of Air Methods Corporation Emergency Medical System interior that provides 3 patient locations to be configured one of two ways using an Intensive Care Transport Litter (ICTL) and Ferno stretchers.</td>
<td>MDL 141-0009 rev B dated 13 June 1993.</td>
<td>1. Single and dual patient configurations are to be installed in strict accordance with Air Methods Corporation Maintenance Manual AM/MS 050-0110 rev N/C dated 15 April 1993 or later approved revision and FAA approved Rotorcraft Flight Manual Supplement RFMS 080-0114 rev N/C date 19 June 1993...</td>
<td>11/01/2007</td>
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<td>EASA.IM.R.S.01249</td>
<td>AAR COMPOSITES, INC.</td>
<td>S-92A</td>
<td>Installation of Partial Search and Rescue Interior Configuration.</td>
<td>Install in accordance with AAR Composites Master Data List SAR-620-010, Revision M dated 1/31/2007, or later approved revisions. Maintenance in accordance with AAR Composites Document Number AAR-ES-1169. Sikorsky S-92A SAR Search and Rescue (SAR) Interior Continued Airworthiness Document, Revision A.</td>
<td>No passengers permitted in the main cabin until the complete interior arrangement has been EASA or cognizant National Airworthiness Authority approved.</td>
<td>07/05/2007</td>
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<td>PARAVION TECHNOLOGY, INC.</td>
<td>BELL 206L-4/BELL 206A/206L-4</td>
<td>Laser Mapping Installation</td>
<td>Master Drawing List DL-206LM-100 rev NC dated March 5, 1999, or later approved revision. Flight Manual Supplement FR-206LM-109 dated April 12, 1999, or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine/ configuration shall be determined by the installer.</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02439NY-D</td>
<td>SIKORSKY S-92A</td>
<td>Chelton Tetra Transceiver Provision Installation in a Sikorsky S92 Helicopter.</td>
<td>Installation in accordance with Keystone Document Reference ID/01234, rev. B or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone Document Reference IC/01234, rev. - or later approved revisions.</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02414NY-D</td>
<td>SIKORSKY S-92A</td>
<td>UIN Sliding Entry Door Installation in a Sikorsky S-92 Helicopter.</td>
<td>Installation in accordance with Keystone Document Reference IDL01270, rev. A or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01270, rev. - or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone...</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02405NY-D</td>
<td>SIKORSKY S-92A</td>
<td>Logo Light Installation in a Sikorsky S-92 Helicopter.</td>
<td>Installation in accordance with Keystone Document Reference IDL01272, rev. - or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM01272, rev. - or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone...</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02407NY-D</td>
<td>SIKORSKY S-92A</td>
<td>Auxiliary DC Power System Installation in a Sikorsky S-92 Helicopter.</td>
<td>Installation in accordance with Keystone Document Reference IDL01299, rev. - or later approved revisions. Operation in accordance with Keystone Rotorcraft Flight Manual Supplement Reference FM-1299, rev. - or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone...</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02121NY-D</td>
<td>SIKORSKY S-92A</td>
<td>Shelf Installation in Sikorsky S-92 Helicopter.</td>
<td>Installation in accordance with Keystone Document Reference IDL01334, rev. - or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone Document Reference IC01334, rev. - or later approved revisions.</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02132NY-D</td>
<td>SIKORSKY S-92A</td>
<td>Ramp Gap Bridge Installation in a Sikorsky S-92 Helicopter.</td>
<td>Installation in accordance with Keystone Document Reference IDL01342, rev. - or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone Document Reference IC01342, rev. - or later approved revisions.</td>
<td>02/05/2007</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02124NY-D</td>
<td>SIKORSKY S-92A</td>
<td>Adlissomable Cabin Window Installation on a Sikorsky S-92 Helicopter.</td>
<td>Installation in accordance with Keystone Document Reference IDL01229, rev. D or later approved revisions. Instructions for Continued Airworthiness in accordance with Keystone Document Reference IC01229, rev. A or later approved revisions.</td>
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<td>DART AEROSPACE Ltd.</td>
<td>REV. 1</td>
<td>EUROCOPTER DEUTSCHLAND EC-135- D135-751 Skidtube and Skidtube Extension Installation iaw FAA STC SR01709SE.</td>
<td>Master Drawing List MDL-D135-751, Revision D Installation Instructions IIN-D135-751, Revision C Instructions for Continued Airworthiness ICA-D135-751, Revision 2</td>
<td>D135-751 Skidtube and Skidtube Extension Installation iaw FAA STC SR01709SE.</td>
<td>10/12/2006</td>
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<td>EASA.IM.R.S.01281</td>
<td>EATON AEROSPACE GLENOLDEN</td>
<td>-</td>
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<td>Installation of Eaton Aerospace/Tedeco Products Zapper pulsed electric chip detector system 1K7707 on Sikorsky Helicopter S-76A and C Models with Turbomeca Arriel 1S, 1S1, 2S1 and 2S2 Engines in accordance with Installation Drawing List IDL00508, Rev. B, dated May 11, 2006 Note: This STC superseded...</td>
<td>1. For the Sikorsky S-76 helicopters fitted with Arriel 1S engines - the Arriel 1S and 1S1 engines must be equipped with TU 208 and TU 232 prior to the installation of the 1K7707/Zapper System. 2. Eaton Aerospace. Rotorcraft Flight Manual Supplement, Document FM 00508, Revision 1, FAA approved May...</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>03/07/2008</td>
<td>EAGLE COPTERS MAINTENANCE Ltd.</td>
<td>TCCA STC SH92-30</td>
<td>BELL 204B</td>
<td>Vertical Reference Door.</td>
<td>A20468DCL1 - Document Control List, Revision 0 dated 28 August 1991 or later approved revision/EA204-68-18 - Service Instruction, Revision 0 dated 29 August 1991 or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; and &quot;Associated Technical Documentation&quot;. Compatibility with other aircraft/Maintenance configurations shall be determined by the installer. This certificate shall remain...</td>
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The table above lists various STCs (Supplemental Type Certificates) issued by the EASA for different aircraft models, including descriptions, associated technical documentation, limitations, and issuance dates. Each entry includes the STC holder, related models, and details about the installation and compatibility requirements.
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<th>EASA STC no.</th>
<th>STC Holder</th>
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<th>Foreign/NAA STC ref.</th>
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<td>EASA.IM.R.S.01291</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02129NY-D</td>
<td>SIKORSKY S-76C, SIKORSKY S-76B, SIKORSKY S-76A</td>
<td>Installation of Keystone Helicopter Corporation “Silencer” Wall panel System. Classified as Basic STC by FAA.</td>
<td>IDL0078 rev – dated 15 February 2007 – Installation Drawing ListIC00916 rev – dated 18 April 2006 – Instructions for Continued Airworthiness</td>
<td>Compliance with FAR 29.93 does not have been established for the installation of this system. An approved emergency lighting system is required. Compliance with FAR 29.783(c)(x), 29.803(a)(b), 29.807(a)(b), 29.809(c), 29.811, 29.813(e), 29.1301(a)(i) and 29.1557(a)(i) has not been established for the insta...</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02131NY-D</td>
<td>SIKORSKY S-76C, SIKORSKY S-76B, SIKORSKY S-76A</td>
<td>Installation of Garmin G1000 Multi Function Display. This STC has been classified as “BASIC” by the FAA and has been approved under the PTVP. The flight manual has been reviewed and is acceptable.</td>
<td>* IDL01217 rev – Installation Drawing List – dated 17 February 2007 or later approved revision* IDL01217 rev – Installation Drawing List – dated 17 February 2007 or later approved revision* IDL01217 rev – Installation Drawing List – dated 17 February 2007 or later approved revision</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA.IM.R.S.01295</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02130NY-D</td>
<td>SIKORSKY S-76C, SIKORSKY S-76B, SIKORSKY S-76A</td>
<td>Installation of Garmin G1000 Multi Function Display. This STC has been classified as &quot;BASIC&quot; by the FAA and has been approved under the PTVP. The flight manual has been reviewed and is acceptable.</td>
<td>IDL01357 rev – dated 11 February 2007 11 February 2007 11 February 2007 11 February 2007</td>
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<td>EASA.IM.R.S.01296</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02123NY-D</td>
<td>SIKORSKY S-76C SIKORSKY S-76B SIKORSKY S-76A</td>
<td>Installation of Provisions for a EuroAvions EuroNav IV System. STC SR02123NY-1-G classified as Basis by the FAA through FAA Order 8110.52.</td>
<td>IDL01318 rev B dated 22 December 2006 – Installation Drawing ListC01318 rev – dated 22 August 2006 – Instructions for Continued Airworthiness</td>
<td>The FAA has not certified, and EASA has not validated, the equipment for which these provisions are intended. Additional approvals are required to install this equipment. The equipment installation must be evaluated to ensure that it complies with the applicable Certification Specifications.</td>
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<td>EASA.IM.R.S.01303</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH06-88</td>
<td>BELL 412 EP BELL 412 CF BELL 214B-1 BELL 214B BELL 212 BELL 206B BELL 205A-1 BELL 205A BELL 204B</td>
<td>Skidtube Installation</td>
<td>For all interested products MDL-D205-434 Revision A dated 13 April 2006 or later approved revision - Master Document List - Bell 204D04-635 Revision C dated 30 March 2006 or later approved revision - Installation Instructions ICA-204-635 Revision 0 dated 30 March 2006 or later approved rev...</td>
<td>1. Operation of the Bell 214B and 214B-1 model types with Dart D205-434-071 skidtubes and the required Dart D212-664-101261 Crossable installation as incorporated by EASA STC No. IM.R.S.01304 (Canadian STC SH94-3) is limited to a maximum take off and landing weight of 1150 lbs. Longitudinal cen...</td>
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<td>EASA.IM.R.S.01304</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH01-6</td>
<td>BELL 412 EP BELL 412 CF BELL 214B-1 BELL 214B BELL 212 BELL 206B BELL 205A-1 BELL 205A BELL 204B</td>
<td>Cross Tube Installation in accordance with Transport Canada/STC SH01-6 Issue 3, dated 8 June 2005. Revision to LBA STCRC 1181 dated 15 January 2002</td>
<td>For Bell 214B and 214B-1 models only FMS- D214-664 Revision A dated 8 June 2005 or later approved revisionMaster Document List MDL- D212-664 revision A dated 8 June 2005 or later approved revisionInstallation Instructions ICA-D214-664 revision 2 dated 17 January 2006 or later approved revision,inv...</td>
<td>Bell 214B and 214B-1 models onlyOperation of the Bell 214B and 214B-1 model types with the cross tube installation is limited to a maximum take off and landing weight of 5403 kg (11,900 lbs). Longitudinal centre of gravity limits have changed accordingly and are provided in Dart Aerospace FMS-D214-6...</td>
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<td>EASA.IM.R.S.01305</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SR09373RC-D</td>
<td>412, 412EP</td>
<td>Bell Helicopter Vibration monitor (BHVM) System</td>
<td>AA-04118 original issue dated 22 June 2005 or later approved revision - Drawing List A, AA-04049, Revision A, dated 14 June, 2005, or later approved revisions - Instructions for Continued Airworthiness. AA-04118 Original issue dated 22 June 2005 or later approved revisions - Flight Manual...</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SR09347RC-D</td>
<td>212, 412, 412 EP</td>
<td>Helicopter Emergency Egress Lighting (Heli) System. This STC is classified as &quot;Basis&quot; by the FAA and is approved under the Post Type Validation Principles (PTVP).</td>
<td>Aeronautical Accessories, Inc. Drawing List Report No. AA-02327, Revision A, dated September 9, 2002, or later FAA/DA/Approved revision instructions for Continued Airworthiness provided in Aeronautical Accessories Inc. Report AA-02329, Revision C, dated 1 July 2005 or later AEG accepted revision A...</td>
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<td>EASA.IM.R.S.01307</td>
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<td>FAA SR01278NY-D</td>
<td>EASA.IM.R.S.01307</td>
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<td>412 EP</td>
<td>L-3 Communications FA230 Series Combined Cockpit Voice Recorder and Flight Data Recorder Kit. This STC is classified as 'BASIC' by the FAA and is approved under the Post Type Validation Principles (PTVP).</td>
<td>AA-05153 Revision H dated 16 June 2006 or later approved revision – Drawing List AA-05153 Revision F dated 16 June 2006 or later approved revision – Installation Report AA-05063 Original Issue dated 27 July 2006 or later approved revision – Flight Manual Supplement AA-05064 Revision A dated 23 June 2006.</td>
<td>No credit is given for operational requirements that require both a CVR and FDR.</td>
<td>04/05/2007</td>
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<td>EASA.IM.R.S.01308</td>
<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>FAA STC SR01946LA</td>
<td>EASA.IM.R.S.01308</td>
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<td>AB139, AW139</td>
<td>Installation of Apical Industries, Inc. Liferaft Kit.</td>
<td>AA-05062 Revision H dated 16 June 2006 or later approved revision – Drawing List AA-05153 Revision F dated 16 June 2006 or later approved revision – Installation Report AA-05063 Original Issue dated 27 July 2006 or later approved revision – Flight Manual Supplement AA-05064 Revision A dated 23 June 2006.</td>
<td>1. This approval requires the inspections and overhaul schedules to be performed as mandated by the Instruction for Continued Airworthiness, ICA139-1, Revision N/C, dated July 27, 2006, or later approved revision.</td>
<td>19/10/2007</td>
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<td>EASA.IM.R.S.01311</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SR01474NY</td>
<td>EASA.IM.R.S.01311</td>
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<td>412 EP</td>
<td>Installation of Honeywell Mark XXII ESPRIS for Bell Helicopter Textron 412EP helicopters. Validation of FAA STC SR01474NY dated January 9, 2007</td>
<td>FAA drawing list report 4123401-MDL01, Revision F, dated 01-03-07, or later approved revision – AAI Rotorcraft Flight Manual Supplement document 4123401-FMS01, Revision C dated 06 June 2007, or later approved revision – Instructions for continued airworthiness report 4123401-RMM01 dated 01-03-2007 or later...</td>
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<td>EASA.IM.R.S.01312</td>
<td>THE NORDAM GROUP LLC</td>
<td>FAA STC SR00548MM / EASA STC 20</td>
<td>EASA.IM.R.S.01312</td>
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<td>369DE/EFF/HYMHE, 500N, 800N</td>
<td>Installation of modified front window canopy. Note: This STC supercedes EASA STC 2004-112.</td>
<td>FAA Instruction 412-2520-022, Revision E, dated May 9, 2006, or later approved revision – AAI Rotorcraft Flight Manual Supplement document 4123401-FMS01, Revision C dated 06 June 2007, or later approved revision – Instructions for continued airworthiness report 4123401-RMM01 dated 01-03-2007 or later...</td>
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<td>SKYTEAC SYSTEMS LTD.</td>
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<td>ASA-04126, rev B</td>
<td>EUROCOPTER FRANCE EC 155B</td>
<td>Installation of a triple litter stacking system</td>
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<td>Installation of a SFM Automatic Flight Control System (AFCS) on Bell 407</td>
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### List of EASA Supplemental Type Certificates

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<td>EASA.IM.R.S.01320</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>REV. 1</td>
<td>FAA STC SA0475SE</td>
<td>BELL 407, BELL 206L-4, BELL 206L-3, BELL 206L-1, BELL 206, BELL 206B, BELL 206A, BELL 206</td>
<td>Replacement of the original landing gear skid tubes and saddles with DART Aerospace Model D206-642 skid tubes and saddles. This TV modifies STC EASA.IM.R.S.01320</td>
<td>MDL-D206-642 revision Q dated 8 March 2007 or later approved revision – Master Drawing List D206-642 revision K dated 29 September 2006 or later approved revision – Installation Instructions/CA-D206-642 revision 0 dated 7 March 2007 or later approved revision – Instructions for continued Airworthiness.</td>
<td>1: This STC is approved only for the production configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer. Installation of the Rear Shoulder Harness on AS350B3 with the modification OP-3369 incorporated is not approved.</td>
<td>10/08/2007</td>
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<td>EASA.IM.R.S.01321</td>
<td>AIRBUS HELICOPTERS CANADA LIMITED</td>
<td>TCCA STC SH97-34 ISSUE 3</td>
<td>AS 355 E/F/F1/F2/N AS 350 B1/B2/B3/B4/D</td>
<td>Installation of Forward Opening Rear Cargo Doors. (Modification approved by TCCA under STC SH97-34 Issue 3).</td>
<td>Definition and installation/ECL Master Drawing List No. MDL-06-002, Issue All and maintenance/ECL Instructions for Continued Airworthiness No. ICA-ECL-141 Rev. 0</td>
<td>1; This STC is approved only for the production configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft/ engine configurations shall be determined by the installer. Installation of the Forward Opening Rear Cargo Doors on AS350B3 with the modification OP-3369 incorporated is not approved.</td>
<td>18/10/2007</td>
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<td>EASA.IM.R.S.01322</td>
<td>RSG PRODUCTS, Inc.D/B/A INTEGRATED FLIGHT SYSTEMS, IN</td>
<td>FAA STC SH947SW</td>
<td>EUROCOPTER FRANCE AS 350F AS 350F1 EUROCOPTER FRANCE AS 350F2 EUROCOPTER FRANCE AS 350F EUROCOPTER FRANCE AS 350E</td>
<td>Installation of freon air conditioner system with single or double condenser blower arrangement.</td>
<td>1. Single blower configuration defined by Drawing List Report DL-8 Revision A or later FAA approved revision. 2. Double blower configuration defined by Drawing List Report DL-8-1 Revision N/C or later FAA approved revision. 3. Flight Manual Supplement dated 14 July 1993 or later FAA approved revision. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>07/07/2008</td>
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<td>EASA.IM.R.S.01327</td>
<td>AIR COMM CORPORATION, L.L.C</td>
<td>FAA STC SH40578M</td>
<td>S-76A, S-76B, S-76C</td>
<td>Installation of bleed air cabin heater and windshield defrost system. Note: STC SH40578M was originally approved by the UK CAA under Airworthiness Approval Note No. 25242. This approval adds models S76A and S76B to the approval. This STC supersedes UK CAA Airworthiness Approval Note No. 25242.</td>
<td>DL-S76H revision I dated 15 December 1995 or later approved revision.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs “Description” and “Associated Technical Documentation” and “Compatibility of this modification with previously approved modifications must be determined by the installer.”</td>
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<td>EASA.IM.R.S.01335</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH62-B</td>
<td>ROBINSON R44 II ROBINSON R44</td>
<td>D044-662 Bearpaw Revision to DGAC STC: RA-D206-628 – Maintenance Instruction DGAC approved 30/04/2003, with addition of R44II aircraft model and larger Bearpaw Kits.</td>
<td>Dart Aerospace Ltd. Installation Instructions RN-D044-662 rev. D, dated 14/02/2007 or later approved revisions.</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs “Description” and “Associated Technical Documentation”. Compatibility of this modification with previously approved modifications must be determined by the installer.</td>
<td>3/05/2007</td>
<td>Active</td>
<td>10116342</td>
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<td>EASA.IM.R.S.01337</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SH8003S-W</td>
<td>SA 365 N, SA 365 N1 AS 365 N2</td>
<td>Installation of combination position lights and strobe lights with supplemental anti-collision lights replacing the red, green and white position lights..</td>
<td>365A73-9062 Revision C dated 12 September 1996 or later approved revision - AEC Drawing List.</td>
<td>Compatibility of this modification with previously approved modifications must be determined by the installer.</td>
<td>13/06/2007</td>
<td>Active</td>
<td>10116343</td>
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<td>EASA.IM.R.S.01338</td>
<td>AIRBUS HELICOPTERS, Inc.</td>
<td>FAA STC SH7016S-W</td>
<td>AS 365 N2, SA 365 N1</td>
<td>Installation of 4.0 inch Footwell in the passenger cabin of the Model SA365A73-9022.</td>
<td>365A73-9022 revision C dated 18 February 1995 or later approved revision - AEC Drawing List.</td>
<td>Compatibility of this modification with previously approved modifications must be determined by the installer.</td>
<td>13/06/2007</td>
<td>Active</td>
<td>10116344</td>
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<td>EASA.IM.R.S.01339</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>TCCA STC SH62-14</td>
<td>BELL 407 BELL 206L BELL 206B BELL 206A</td>
<td>D206-628 Hell Access Steps. This STC has been classified as “BASIC” by the FAA and is approved under the Type Validation Principles.</td>
<td>DLL-D206-628 – Maintenance Manual Supplement. Revision A dated 27 November 1997 or later approved revision.</td>
<td>Compatibility of this modification with previously approved modifications must be determined by the installer.</td>
<td>26/07/2007</td>
<td>Active</td>
<td>10116345</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>FAA STC</td>
<td>Rev.</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.IM.R.S.01341</td>
<td>APICAL INDUSTRIES, Inc.d/b/a DART AEROSPACE</td>
<td>FAA STC SH76555W</td>
<td>BO 105 C/S</td>
<td>Replacement of the MBB Oil Cooler Fairing, MBB P/N 105-61802</td>
<td>FAA STC SH 76555W</td>
<td>Installation in accordance with ERA Drawing 10579-100-001, dated September 27, 1989, or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>15/08/2007</td>
<td>Active</td>
<td>10016947</td>
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<td>EASA.IM.R.S.01345</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA STC SR00280DE</td>
<td>BELL 407</td>
<td>Installation of Bleed Air Cabin Heater System.</td>
<td>DL-407H-100 revision D dated 24 February 1997 or later approved Revision – Master Drawing List.PR-407H-100M revision 2 dated 18 April 2002 or later approved revision – Rotorcraft Flight Manual Supplement PR0407H-120 M revision NC dated 14 April 1997 or later approved revision – Instructions for Con...</td>
<td>21/08/2007</td>
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<td>EASA.IM.R.S.01346</td>
<td>ISOLAIR Inc., ISOLAIR HELICOPTER SY</td>
<td>FAA STC SH4768NM</td>
<td>BELL 206B</td>
<td>Installation of Isolair 3900-206 Spray System consisting of a fibreglass 130 gallon belly tank and spray booms for agricultural applications.</td>
<td>39206INS revision B dated 22 August 2006 or later approved revision – Installation Manual; 3900-206 revision IR dated 14August 2006 or later approved revision – Master Drawing List.3900CACA revision A dated 6 August 2006 or later approved revision – Maintenance Instructions;39206INS revision B dated...</td>
<td>06/03/2008</td>
<td>Active</td>
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<td>EASA.IM.R.S.01348</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA STC SR000280DE</td>
<td>BELL 206A/B/L-1L-3/4-L, 407</td>
<td>Installation of an Equipment Tray.</td>
<td>DL-206V407ET-100 revision NUC dated 27 July 1999 or later approved Revision – Master Drawing List</td>
<td>This approval is for a shelf only. The shelf is rated for a 15 pound load and must be placarded as shown in drawing 206ET-1000, Revision B dated 9 February 1993 or later approved revision. On 206A and B models, this approval is for serial numbers 154 and subsequent due to airframe differences.</td>
<td>21/08/2007</td>
<td>Active</td>
<td>10016952</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA/IM.R.01350</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA STD-SR0034DE</td>
<td>BELL 407</td>
<td>Installation of External Mirror. (This STC has been classified as ‘BASIC’ by the FAA in accordance with FAA Order 8130.52 and has been approved using the PTP Procedure.)</td>
<td>DL-407EM-100 Revisions NCA dated 9 January 1998 or later approved Revisions – Master Drawing List.</td>
<td>06/09/2007</td>
<td>Active</td>
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<td>EASA/IM.R.01351</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA STD-SR0001DE</td>
<td>BELL 206B/L/L-1L-3L-4, 407</td>
<td>Provision for Mount of a Video Monitor Placed on Left or Right Instrument Panel Console.</td>
<td>DL-206/407MM-110 Revision N/C dated 27 July 1999 or later approved Revision – Master Drawing List.</td>
<td>21/08/2007</td>
<td>Active</td>
<td>10101955</td>
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<td>EASA/IM.R.01352</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA STD-SR0048DE</td>
<td>BELL 407</td>
<td>Installation of mounting hardware and approval for the aircraft to be operated by a single pilot from the left crew seat.</td>
<td>DL-407LS-100 Revision N/C dated 1 April 2002 or later approved Revision – Master Drawing List PR-407LS-100M Revision 1 dated 3 March 2003 or later approved revision – Rotorcraft Flight Manual Supplement PR-407S-120 M Revision 3 dated 5 December 2002 or later approved Revision – Instructions for Continued Airworthiness.</td>
<td>21/08/2007</td>
<td>Active</td>
<td>10101956</td>
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<td>EASA/IM.R.01361</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC-SH0737</td>
<td>BELL 412 EP BELL 412 CF BELL 412 BELL 212 BELL 206B BELL 205A-1 BELL 205A</td>
<td>DA142-760 Cabin Floor Protectors.</td>
<td>MDL-DA142-760 – Master Drawing List. Revision A dated 29 June 2007 or later approved revision (M) DA142-760 – Installation Instruction, Revision A dated 8 June 2007 or later approved revision (CA) DA142-760 – Maintenance Manual Supplement, Revision 0 dated 29 June 2007 or later approved revision</td>
<td>06/09/2007</td>
<td>Active</td>
<td>10101962</td>
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<td>EASA STC no.</td>
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<td>Model</td>
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<td>Limitations/Conditions</td>
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<td>EASA.IM.R.S.01363</td>
<td>LIFEPORT, LLC</td>
<td>FAA STC SR02054LA</td>
<td>AGUSTA AW-139</td>
<td>Installation of an Emergency Medical System</td>
<td>Master Drawing No. MDL-831-100, Revision 2 dated June 12, 2007 or later FAA approved revision.</td>
<td>1. The installer of this STC must ensure that, if the life vests have been removed with the seats, a life vest for the occupant of the litter system is installed and available to the occupant.</td>
<td>19/02/2007</td>
<td>Active</td>
<td>1016593</td>
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<td>EASA.IM.R.S.01365</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA STC SR00074DE</td>
<td>BELL 407</td>
<td>Installation of 407FL-100 FADEC/Check instrument auxiliary annunciation system. (This STC has been classified as &quot;BASIC&quot; by the FAA in accordance with FAA Order 8110.52 and has been approved using the PTVP Procedures.)</td>
<td>DL-407FL-100 revision A dated 7 September 2004 or later approved revision – Master Drawing List.PR-407FL-100M revision Original dated 26 January 2005 or later approved revision – Rotorcraft Flight Manual Supplement PR-407FL-120 M revision dated 7 September 2004 or later approved Revision – Instruction...</td>
<td>06/02/2007</td>
<td>Active</td>
<td>1016594</td>
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<td>EASA.IM.R.S.01366</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA STC SR00034DE</td>
<td>BELL 407</td>
<td>Infrared Imaging System Installation. This STC has been classified as &quot;BASIC&quot; by the FAA and has been approved under the PTVP.</td>
<td>DL-407SX-100 revision A dated 27 April 1998 or later approved revision – Master Document List.PR-407SX-100M revision 0 dated 26 June 1998 or later approved revision – Flight Manual Supplement.</td>
<td>19/03/2008</td>
<td>Active</td>
<td>1016595</td>
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<td>EASA.IM.R.S.01367</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA STC SR00034DE</td>
<td>BELL 407</td>
<td>Infrared Imaging System Installation. This STC has been classified as &quot;BASIC&quot; by the FAA and has been approved under the PTVP.</td>
<td>DL-407IR-100 revision A dated 21 May 1998 or later approved revision – Master Document List.PR-407IR-100M revision 4 dated 8 July 2005 or later approved revision – Rotorcraft Flight Manual Supplement.</td>
<td>None.</td>
<td>22/05/2008</td>
<td>Active</td>
<td>1016596</td>
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<td>EASA.IM.R.S.01368</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SR00778AT</td>
<td>206L-3, 206L-4</td>
<td>Expanded Instrument Panel. This STC is classified as &quot;BASIC&quot; by the FAA and is approved under the PTVP.</td>
<td>AA-95023 - Drawing List, revision E, dated 23 February 2006 or later approved revision.</td>
<td>16/11/2007</td>
<td>Active</td>
<td>1016597</td>
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<td>EASA.IM.R.S.01369</td>
<td>ACROHELI PRO GLOBAL SERVICES, INC.</td>
<td>FAA STC SR002292AV</td>
<td>BELL 206L-4</td>
<td>Sagem ICS and Crossbow AHRS Installation. Consists of two 8 inch integrated cockpit display units (one PFD and one MFD) including engine instruments and an attitude and heading reference system. This STC is classified as &quot;BASIC&quot; by the FAA and is approved under the Post Type Validation Principles (PTVP).</td>
<td>HP 206-34-33-102 – Certification Data List – revision A dated 3 July 2007 or later approved revision.</td>
<td>Maximum ambient operating temperature of 40 degrees Celsius (104 degrees Fahrenheit).</td>
<td>31/10/2007</td>
<td>Active</td>
<td>1016598</td>
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<td>EASA.IM.R.S.01370</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SH06185O</td>
<td>206L-3, 206L-4</td>
<td>Lower Console Assembly. This STC is classified as &quot;BASIC&quot; by the FAA and is approved under the Post Type Validation Principles (PTVP).</td>
<td>AA-88994 - Drawing List, no revision, dated 8 August 1989 or later approved revision.</td>
<td>09/10/2007</td>
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<td>1016599</td>
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<td>EASA STC no.</td>
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<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>EASA.IM.R.S.01371</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SR01075AT</td>
<td>407</td>
<td></td>
<td>Crew and Passenger Wedge Windows.</td>
<td>AA-9653 - Drawing List, no revision, dated 16 May 1996 or later approved revision AA-9655 - Installation Instructions, no revision, dated 18 May 1996 or later approved revision AA-9655 - Instructions for continued Airworthiness, no revision, dated 16 April 1996 or later approved revision.</td>
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<td>02/10/2007</td>
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<td>10016970</td>
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<td>EASA.IM.R.S.01372</td>
<td>APICAL INDUSTRIES, Inc./d/b/a DART AEROSPACE</td>
<td>FAA STC SR02065LA</td>
<td>AS 355 E/F/F1/F2/N</td>
<td>355-31 Lower Front Fairing Assembly.</td>
<td>Instructions for Continued Airworthiness, ICA350-31, Revision B, dated 3rd January 2007, or later approved revision.</td>
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<td>23/01/2009</td>
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<td>EASA.IM.R.S.01375</td>
<td>GENEVA AVIATION Inc.</td>
<td>FAA STC SR02425SE</td>
<td>EUROCOPTER FRANCE AS 350 D1 EUROCOPTER FRANCE AS 350 C EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 B3 EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B</td>
<td>Installation of a new instrument panel and pedestal per MDL GA162</td>
<td>Master Drawing List GA162 at Rev. Level B dated 11/15/96 or later EASA approved revisions Installation Instructions GA162-2 Rev. dated 8/23/95 or later EASA approved revisions Compliance Checklist CCL-162 Rev. dated August 14, 2007 or later EASA approved revisions</td>
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<td>11/12/2006</td>
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<td>EASA.IM.R.S.01376</td>
<td>GENEVA AVIATION Inc.</td>
<td>FAA STC SH5119 NM</td>
<td>EUROCOPTER FRANCE AS 350N EUROCOPTER FRANCE AS 359F2 EUROCOPTER FRANCE AS 359F1 EUROCOPTER FRANCE AS 359F EUROCOPTER FRANCE AS 359E EUROCOPTER FRANCE AS 359 D EUROCOPTER FRANCE AS 359 C EUROCOPTER FRANCE AS 359 BA EUROCOPTER FRANCE AS 359 B3 EUROCOPTER FRANCE AS 359 B2 EUROCOPTER FRANCE AS 359 B1 EUROCOPTER FRANCE AS 359 B</td>
<td>Rear Bench Seat Modification per MDL GA141</td>
<td>Master Drawing List GA141 Rev. C dated 3/22/95 or later approved revisionsRotorcraft Flight Manual Supplement FLMS 135-110, Rev. B, dated 15.03.2007 or later approved revisionsInstallations and Overhaul Manual Supplement II, Rev. B, dated 15.03.2007</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>03/04/2008</td>
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<td>10016975</td>
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<td>EASA.IM.R.S.01377</td>
<td>METRO AVIATION, Inc.</td>
<td>FAA STC SR00240RC</td>
<td>EUROCOPTER DEUTSCHLAND EC-135</td>
<td>Installation of landing gear skid extensions</td>
<td>Flight Manual Supplement FLMS 135-110, Rev. B, dated 15.03.2007 or later approved revision Installations and Overhaul Manual Supplement II, Rev. B, dated 15.03.2007 or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td></td>
<td>26/11/2007</td>
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<td>10016976</td>
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<td>STC Holder</td>
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<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>System Record</td>
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<td>EAGLE COPTERS MAINTENANCE LTD.</td>
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<td>TC1A STC S1943-37</td>
<td>BELL 204B</td>
<td>DCL 361 Tailboom Vertical Fin Spar Reinforcement. New STC for modification of tailboom vertical fin spar to comply with requirements of FAA AD 99-17-03.</td>
<td></td>
<td>DCL361 - Aero Design Ltd. Document Control List - revision 5 dated 10 October 2000 or later approved revision/MS381-03 - Eagle Copters Ltd. Maintenance Manual Supplement - revision 2 dated 10 October 2000 or later approved revision.</td>
<td>This approval supports an Alternative Means of Compliance (AMOC) for satisfying FAA AD99-17-03. An EASA AMOC authority is required in addition to this STC.</td>
<td>15/11/2007</td>
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<td>LIFEPORT, LLC</td>
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<td>FAA STC SR009602RC</td>
<td>BELL 230</td>
<td>Air medical System. This STC has been classified as &quot;BASIC&quot; by the FAA and has been approved under the PTVP.</td>
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<td>05/03/2008</td>
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<td>EAGLE COPTERS MAINTENANCE LTD.</td>
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<td>FAA STC SR00380DE</td>
<td>BELL 230</td>
<td>Medical System. This STC has been classified as &quot;BASIC&quot; by the FAA and has been approved under the PTVP.</td>
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<td>05/03/2008</td>
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<td>EAGLE COPTERS MAINTENANCE LTD.</td>
<td>REV. 1</td>
<td>TC1A STC S1943-14</td>
<td>BELL 412</td>
<td>D412-758 GPS receiver Gimbal Mount</td>
<td></td>
<td>1) Master Document List MDL-D412-758 Revision D dated 8 April 20082) Installation Instructions HN-D412-758 Revision B dated 8 April 20083) Installation Instructions for Continued Airworthiness ICA-D412-758 Revision 2 dated 7 April 2008</td>
<td>1) Eligible GPS receivers are listed in Installation Instructions HN-D412-758 Revision B dated 8 April 20082) GPS operation is limited to VFR use only</td>
<td>21/04/2008</td>
<td>Active</td>
<td>10115800</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td></td>
<td>FAA STC SR02417NY-D</td>
<td>SKORSKY S-76C</td>
<td>Installation of Honeywell KHF 1050 HF Communication System.</td>
<td></td>
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<td>04/12/2007</td>
<td>Active</td>
<td>10115801</td>
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<tr>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>REV. 1</td>
<td>FAA STC SR02120NY-D</td>
<td>SKORSKY S-92A</td>
<td>Installation of a Cockpit Pressure Refueling Control in a Sikorsky S-92A.</td>
<td></td>
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<td>08/11/2008</td>
<td>Active</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
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<td>FAA STC SR02420NY-D</td>
<td>SKORSKY S-76C</td>
<td>Installation of a Replacement Concorde Lead Acid Battery.</td>
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<td>05/02/2008</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
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<td>FAA SR0070NY</td>
<td>407</td>
<td>Installation of SFIM Force Trim System on Bell 407 helicopters. This STC is classified Basic by the FAA and is approved under the cost type Validation Principles (PTVP)</td>
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<td>07/04/2008</td>
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**EASA STC No.**: SR00360DE

**Certification basis**: FAA STC

**Installation of SFIM Force Trim System**

**Active:**

**Limitations/Conditions**: This STC is classified Basic by the FAA and is approved under the cost type Validation Principles (PTVP)

**Date of Issue**: 07/04/2008

**Status**: Active

**System Record**: 10115804
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<th>List of EASA Supplemental Type Certificates</th>
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<td><strong>EASA/IM.R.S.01391</strong></td>
<td><strong>BELL HELICOPTER TEXTRON, Inc.</strong></td>
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<td>Model</td>
<td>Rubber mounted replacement chin window. This STC is classified as &quot;BASIC&quot; by the FAA and is approved under the Post Type Validation Principles (PTVP).</td>
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<td>Certification basis</td>
<td>AA-60122-D Drawing List, revision H, dated 1 May 2007 or later approved revision AA-60156 – Installation Instructions, revision J, dated 26 April 2007 or later approved revision; oAIA-03112 – Installation Instructions, revision C, dated 1 May 2007 or later approved revisionAA-60159 – Installation Instructions, revision D.</td>
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<td>Status</td>
<td>FAA STC</td>
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| **EASA/IM.R.S.01393** | **BRISTOL AEROSPACE LTD.** |
| FAA STC | SRO24916 |
| Rev. | |
| Model | TCCA STC SH91-15; FAA STC SH26 |
| Certification basis | Installation of Wire Strike Protection System. |
| Limitations/Conditions | The installation is to be done in accordance with Bristol Aerospace Ltd. Drawing List EM103457 Issue E, DOT approved June 12, 2001 or later DOT approved Revision. |
| Date of Issue | 10/03/2008 |
| Status | Active |

| **EASA/IM.R.S.01395** | **AIRWOLF AEROSPACE L.L.C.** |
| FAA STC | SRO24917CH |
| Rev. | R44, R44 II R22, R22 ALPHA/BETA/ MARINER |
| Model | Airwolf Rotor Blade Protective Tape Installation of a leading edge protective tape for Robinson R22 and R44 helicopter models according to FAA STC SR02491CH Note: The recommendation for approval is given based on a review performed in accordance with the Type Validation/Post Type Validation Principles. |
| Certification basis | Installation instructions: Airwolf Aerospace, LLC Installation Manual RPBT-100, Rev. A dated October 20, 2007 or later FAA approved revisions. |
| Limitations/Conditions | 07/02/2008 |
| Date of Issue | Active |
| Status | FAA STC | |
| System | Active |

| **EASA/IM.R.S.01396** | **ONBOARD SYSTEMS INTERNATIONAL** |
| FAA STC | SRO04075E |
| Rev. | REV. 1 |
| Model | MD HELICOPTERS 500N MD HELICOPTERS 369HM MD HELICOPTERS 369HE MD HELICOPTERS 369FF MD HELICOPTERS 369F MD HELICOPTERS 369E MD HELICOPTERS 369D |
| Certification basis | Onboard Systems Model 200-187-00 cargo hook on helicopters 369D, 369E, 369F, 369HE, 369HM, 369HS, 500N/EASA validation of FAA STC SR04075E; amended 2 October 2007, or later approved 07This STC is classified as 'Basic' by the FAA and is approved under the post type Validation Principles (PTVP). |
| Limitations/Conditions | Fabrication of kit in accordance with FAA Approved Onboard Systems Master Drawing List No. 155-030-00, Revision 12, dated February 22, 2008, or later FAA approved revisionInstallation/Installation of this replacement cargo hook in accordance with FAA approved Onboard Systems Owner's Manual No. 120-047-00, Revia... Approval of this change in type design applies to only those McDonnell Douglas cargo hook suspension systems and Breeze-Eastern cargo hooks shown in the following table:A copy of this C... |
| Date of Issue | 18/03/2008 |
| Status | Active |

| **EASA/IM.R.S.01398** | **DART AEROSPACE Ltd.** |
| FAA STC | SRO24920 |
| Rev. | D205-576 Rev D |
| Model | TCCA STC SH92-20; ISSUE 2; FAA |
|
| Limitations/Conditions | Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third person(s)... |
| Date of Issue | 03/06/2008 |
| Status | FAA STC | |
| System | Active |

| **EASA/IM.R.S.01402** | **ONBOARD SYSTEMS INTERNATIONAL** |
| FAA STC | SRO17856E |
| Rev. | |
| Model | EUROCOPTER FRANCE EC 120 B |
| Certification basis | Installation of Cargo Hook Suspension Kit PN 200-308 - 00. |
| Limitations/Conditions | 1. Definition and Installation in accordance with - Onboard Systems Model 200-308-00 Cargo Hook Suspension Kit - Onboard Systems Master Drawing List 155-115-00, Revision 1, dated October 22, 2007 or later approved Revision Onboard Systems Owner's Manual 120-123-00, Revision 1, dated Octo...
1. All limitations of RPM supplement apply. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The... |
| Date of Issue | 22/01/2008 |
| Status | Active |

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Note: The table contains a list of EASA Supplemental Type Certificates, including details such as FAA STC numbers, models, certification basis, limitations, and dates of issue. The documents are available at easa.europa.eu.
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<th>EASA STC no.</th>
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<th>Rev. level</th>
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<th>Models</th>
<th>Description</th>
<th>Certification basis</th>
<th>Associated Technical Documentation</th>
<th>Limitations/Conditions</th>
<th>Date of Issue</th>
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<td>ASI SERVICES</td>
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<td>FAA SH296NM</td>
<td>BELL 206L-1</td>
<td>Installation of a Rolls Royce model 250-C30P engine and associated airframe and tail rotor control modifications similar to the Model 206L-3. This STC has been classified as &quot;BASIC&quot; by the FAA and has been approved under the PTVP.</td>
<td>Air Services International FAA Sealed Master Drawing List ASI-206L-30P revision 8 dated 20 June 2002 or later approved revision – Air Services International Installation Manual ASI-206L-1-FM revision 15 dated 3 January 2006 or later approved revision – Rotorcraft Flight Manual</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by th...</td>
<td>09/06/2008</td>
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<td>EASA.IM.R.S.01404</td>
<td>DIAMOND J. Inc.</td>
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<td>FAA STC NO. SR00031LB DATED MA</td>
<td>BELL 206L-4 BELL 206L-3 BELL 206L-1 BELL 206L</td>
<td>Installation of Analog/Digital Turbine Outlet Temperature (TOT) Indicator with Exceedance logging Capability. This STC has been classified as &quot;Basic&quot; by FAA and has been approved under the PTVP.</td>
<td>See attached sheet.</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>REV. 1</td>
<td>FAA STC SR02439NY-D</td>
<td>SIKORSKY S-92A</td>
<td>Installation of a High Intensity Strobe Lights. This STC has been classified as BASIC by the FAA and has been approved under the PTVP.</td>
<td>FM01368 revision dated 24 January 2008 or later approved revision Flight manual Supplement LIC01368 revision dated 20 November 2006 or later approved revision Instructions for Continued Airworthiness/DL 01368 revision A dated 11 January 2008 or later approved revision Installation Drawing List</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>DART AEROSPACE USA, Inc.</td>
<td>FAA STC SR00211SE</td>
<td>EUROCOPTER FRANCE AS 350F2 EUROCOPTER FRANCE AS 350F1 EUROCOPTER FRANCE AS 350F EUROCOPTER FRANCE AS 350E EUROCOPTER FRANCE AS 350B1</td>
<td>Installation of a two place front seat.</td>
<td>D350-764 Shoulder Harness Kit. This STC has been classified as &quot;BASIC&quot; by the FAA and has been approved under the PTVP.</td>
<td>GA 150 Rev A dated 14 June 1995 or later approved revision - Geneva Aviation Master Drawing ListMDL-D350-764 Revision C dated 18 December 2007 or later approved revision - Dart Aerospace USA Master Document ListMDL-D350-764 revision B dated 16 October 2007 or later approved revision - Dart Aero...</td>
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<td>EASA.IM.R.S.01417</td>
<td>PRATT &amp; WHITNEY</td>
<td>FAA STC SR00125BO</td>
<td>MD HELICOPTERS 500N MD HELICOPTERS 369H MD HELICOPTERS 369HM MD HELICOPTERS 369HE MD HELICOPTERS 369H MD HELICOPTERS 369F MD HELICOPTERS 369F2 MD HELICOPTERS 369E MD HELICOPTERS 369D MD HELICOPTERS 369C MD HELICOPTERS 369B MD HELICOPTERS 369A</td>
<td>Installation of a Pratt &amp; Whitney Engine Services, Inc.-Honeywell IntelliStart Plus +TM Electronic Control System P/N DPU-K-010Note: This STC reflects the transfer from Altair Avionics Corporation to Pratt &amp; Whitney Engine Services, Inc.</td>
<td>Pratt &amp; Whitney Engine Services, Inc-Honeywell InstallationManual DPU-I-010-3, Revision A, dated July 18, 2000, or later approved revisions.</td>
<td>1. Aircraft must have Engine modified in accordance withEASA STC EASA.IM.R.S.01019, FAA STC SE00027EN. 2. Pratt &amp; Whitney Engine Services, Inc. Rotorcraft Flight Manual DPPU-T-250-2 dated July 27, 2000 or later approved revision, is required.3. This system is designed as an electronic aid f...</td>
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<td>Associated Technical Documentation</td>
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<td>PRATT &amp; WHITNEY</td>
<td>FAA STC SR001680, LBA RC1237</td>
<td>REV. 1</td>
<td>EUROCOPTER FRANCE AS 355N, EUROCOPTER FRANCE AS 355F1, EUROCOPTER FRANCE AS 355F, EUROCOPTER FRANCE AS 355D, EUROCOPTER FRANCE AS 350 B, EUROCOPTER FRANCE AS 350 B3, EUROCOPTER FRANCE AS 350 B2, EUROCOPTER FRANCE AS 350 B1, EUROCOPTER FRANCE AS 350 B, EUROCOPTER DEUTSCHLAND BO-105D, EUROCOPTER DEUTSCHLAND BO-105</td>
<td>Installation of Pratt &amp; Whitney Engine Services, Inc. SmartCycle+ monitoring and recording system</td>
<td>AML SR001690</td>
<td>1. Required maintenance, as prescribed in the applicable airframe or engine Instructions for Continued Airworthiness, shall not be predicted on the data obtained from the Pratt &amp; Whitney Engine Services, Inc. monitoring/recording system.2. This system shall not be used to supersede or replace the p...</td>
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<td>PRATT &amp; WHITNEY</td>
<td>FAA STC SR001280, LBA RC 1</td>
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<td>BELL 206A-1</td>
<td>Installation of a Pratt &amp; Whitney Engine Services, Inc. SmartCycle Plus System P/N EMU-K-010-2</td>
<td>Pratt &amp; Whitney Engine Services, Inc. Installation Manual, EMU-K-010-2, Revision B, dated August 22, 2000 or later approved revisions</td>
<td>1. Pratt &amp; Whitney Engine Services, Inc. Rotorcraft Flight Manual EMU-T-250-1, dated August 31, 2000 or later approved revision, is required. This system is designed as an electronic aid for recording engine trend and cycle data. This system does not eliminate any scheduled maintenance or parts rep...</td>
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<td>PRATT &amp; WHITNEY</td>
<td>FAA STC ST0004860, LBA RC 1048</td>
<td>ROBINSON R22 BETA ROBINSON R22</td>
<td>Installation of a Pratt &amp; Whitney Engine Services, Inc. Health Usage Monitoring System (HUMS) Model 1522, P/N HUMS- A.030-1</td>
<td>HUMS Installation Instructions, HUMS- A.030-1, Version 1.3, dated December 8, 1995, or later approved revisions</td>
<td>1. Federal Aviation Administration approval of the HUMS installation was determined on a non-interference basis. Operation of maintenance must not be predicated on the HUMS.</td>
<td>04/03/2008</td>
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<td>AEROMETALS INC.</td>
<td>FAA STC SR009611A</td>
<td>369HM, 369HS, 500N 369D, 369E, 369F, 369HE</td>
<td>Shut Assy - Landing Gear, p/n 369X0601. EASA validation of FAA approved PMA part.</td>
<td>Instructions for Continued Airworthiness 369XH0601 Rev 03/18/03 or later FAA approved revision. Master Drawing List No. 9 Rev 07/19/03 or later FAA approved revision.</td>
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<td>AEROMETALS INC.</td>
<td>FAA STC SR010621A</td>
<td>369HM, 369HS, 500N 369A, 369D, 369E, 369HE</td>
<td>Exhaust Tailpipe Outlet hanger Assys., p/n 369X06206. EASA validation of FAA approved PMA part.</td>
<td>Aerometals Instructions for Continued Airworthiness AM-18 Rev C 09/11/03 or later FAA approved revision Master DrawingList No. 15 Rev C 03/02/02 or later FAA approved revision.</td>
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<td>EASA.IM.R.S.01433</td>
<td>AEROMETALS INC.</td>
<td>FAA STC ST015050E</td>
<td>369D, 369E, 369F, 500N</td>
<td>Scavenger Air Elbow Assy., p/n AB139, AW139. EASA validation of FAA approved PMA part.</td>
<td>Installation Instructions AMC 66-0403 Rev R 10/05/04 or later FAA approved revision Master DrawingList MDL- AMC-079 Rev R 10/05/04 or later FAA approved revision.</td>
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<td>APICAL INDUSTRIES, Inc dba DART AEROSPACE</td>
<td>FAA STC SR021811A</td>
<td>AB139, AW139</td>
<td>Nose Landing Gear Doors Kit on Agusta AB139, AW139.</td>
<td>Master Document List MD139-2, Revision B, dated November 8,2007, or later approved revision Instructions for ContinuedAirworthiness, ICA139-2, Revision A, dated August 6, 2008, or later approved revision.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA</td>
<td>SH8713SW</td>
<td>230</td>
<td>Replacement of the cabin door with right hand sliding door. This STC has been classified as ‘BASIC’ by the FAA and has been approved under the PTVP.</td>
<td>230-530-102 Revision A dated 28 April 1994 or later approved revision – Sliding Door Installation R.H., 230-530-116 Revision IR dated 1 August 1994 or later approved revision – Sliding Door Installation L.H., BHT-230-08M-24 Revision 0 dated 29 September 1995 or later approved revision – Flight Manual S...</td>
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<td>APICAL INDUSTRIES, Inc.</td>
<td>FAA</td>
<td>SRI01439LA</td>
<td>407</td>
<td>Emergency Float Assemblies.</td>
<td>MD407-1 Revision C dated 5 February 2003 or later approved revisions – Master Document List IC4407-1 Revision N/C dated 12 June 2002 or later approved revision – Instructions for Continued Airworthiness</td>
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<td>APICAL INDUSTRIES, Inc.</td>
<td>FAA</td>
<td>SRI01392LA</td>
<td>A119, AV119MKII</td>
<td>Emergency Float Kits FAA STC 407 dated 21 December 2007.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>FAA STC</td>
<td>SRI018156E</td>
<td>EUROCOPTER FRANCE AS 130 B</td>
<td>Installation of Model 200-321-0 Cargo Hook Swing Suspension Kit and Model 200-322-0 Cargo Hook Sling Suspension Kit.</td>
<td>Definition and installation: 1. Definition and installation: a. Onboard Systems Master Drawing List No. 155-21-00, Rev. 1 dated February 6, 2008 or later EASA approved revisions a. Onboard Systems Owner’s Manual No. 150-131-00 Rev. 1 dated February 4, 2008 or later EASA approved revision a...</td>
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<td>Limitations/Conditions</td>
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<td>FAA STC SH3754SWS</td>
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<td>Installation of DeVore forward facing recognition lights</td>
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<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>01/07/2008</td>
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<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>FAA STC SR01812SE</td>
<td>EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 B2 EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B</td>
<td>Installation of Cargo Hook Kit P/N 200-297-00 and installation of Cargo Hook Kit P/N 200-298-00 This STC is the validation of FAA STC SR01812SE</td>
<td>Definition and installation Onboard Systems Master Drawing List Doc. No. 155-110-00; Rev. 1, dated September 27, 2006, or later, FAA approved revision Onboard Systems Owner’s Manual No. 120-115-00; Rev. 0 dated April 25, 2005, or later FAA approved revisions (for Cargo Hook Kit P/N 200-297-00). Onboard Systems Master Drawing List Doc. No. 155-110-00, Rev. 1, dated September 27, 2006, or later, FAA approved revision</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>21/11/2008</td>
<td>Active</td>
<td>10017027</td>
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<td>EASA.IM.R.S.01448</td>
<td>HERITAGE AVIATION LIMITED</td>
<td>FAA STC SH7333SWS</td>
<td>MD HELICOPTERS 600N MD HELICOPTERS 500N MD HELICOPTERS 369FF MD HELICOPTERS 366E</td>
<td>Installation of enlarged instrument panel and slant console. This STC has been classified as ‘BASIC’ by the FAA and has been approved under the PTVP.</td>
<td>A00-13003 revision B dated 14 June 2007 or later approved revision – Master Drawing List; A09-63001 revision Original dated 2 July 2007 or later approved revision – Flight manual Supplement; A04-73003 revision Original dated 22 March 2007 or later approved revision – Instructions for Continued Airworthiness; A09-63001 revision Original dated 2 July 2007 or later approved revision – Flight manual Supplement; A04-73003 revision Original dated 22 March 2007 or later approved revision – Instructions for Continued Airworthiness;</td>
<td>Not compatible with the MDHI optional centre seat installation.</td>
<td>01/04/2008</td>
<td>Active</td>
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<td>EASA.IM.R.S.01449</td>
<td>DART AEROSPACE Ltd.</td>
<td>TC STC SH05-45</td>
<td>BELL 206B</td>
<td>D206-746 Shroud Air Flow Heater System.</td>
<td>IIN-D206-746 Revision A dated 18 May 2006 or later approved revision – Installation Instructions; MMS-D206-746 Revision A dated 16 March 2006 or later approved revision – Maintenance Manual Supplement; FMS...</td>
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<td>16/04/2008</td>
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<td>EASA.IM.R.S.01451</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH59-221 ISSUE 1</td>
<td>EUROCOPTER FRANCE EC 120 B</td>
<td>D120-638 Beeswape Installation.</td>
<td></td>
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<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification b...</td>
<td>13/08/2008</td>
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<td>EASA.IM.R.S.01452</td>
<td>DART AEROSPACE Ltd.</td>
<td></td>
<td></td>
<td>EUROCOPTER FRANCE EC 120 B</td>
<td>D120-639 Wearplate Installation.</td>
<td>L: Determination and Installation in accordance with Dart Aerospace Ltd. Installation instructions IIN-D120-639, revision A, dated December 11, 2000, or later approved revision. Inspection and Maintenance in accordance with: Dart Aerospace Ltd. Instruction for Continued Airworthiness (CA)-D120-63...</td>
<td>None</td>
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<td>EASA.IM.R.S.01456</td>
<td>DART AEROSPACE Ltd.</td>
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<td>EUROCOPTER FRANCE AS 355P</td>
<td>D350-766 interior Trim Kit Installations</td>
<td>IIN-D350-766 revision A dated 3 March 2008 or later approved revision – Installation Instructions;MDL-D350-766 revisionA dated 3 March 2008 or later approved revision – Master Document List;ICA-D350-766 revision 0 dated 3 March 2008 or later approved revision – Instructions for Continued Airworthiness...</td>
<td>None</td>
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<td>EASA.IM.R.S.01458</td>
<td>DART AEROSPACE USA, Inc.</td>
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<td>AGUSTA AW-119 MKI</td>
<td>FFA STC SR00948SE 'D119-675 Hel-Access-Step'.</td>
<td>FDC/aerofilter Rotorcraft Flight Manual Supplement 1076-1200-105 for S-76A +A+++C+ FDC/aerofilter Rotorcraft Flight Manual Supplement 1076-1200-101 for S-76C +FDC/aerofilter 'Documentation Package' 1076-SERIES-DPD-1 Rev IfRor subsequent, including 'Installation Instructions' 1076-I...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by IN...</td>
<td>18/04/2008</td>
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<td>EASA.IM.R.S.01459</td>
<td>FILTRATION DEVELOPMENT CORPORATION</td>
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<td>U. SIKORSKY S-76C+</td>
<td>Installation of an Engine Air Filter System for Sikorsky S-76 models, A+, A++ and C &amp; C+</td>
<td>- FDC/aerofilter Rotorcraft Flight Manual Supplement 1076-1200-105 for S-76A +A+++C+ FDC/aerofilter Rotorcraft Flight Manual Supplement 1076-1200-101 for S-76C +FDC/aerofilter 'Documentation Package' 1076-SERIES-DPD-1 Rev IfRor subsequent, including 'Installation Instructions' 1076-I...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by IN...</td>
<td>05/02/2009</td>
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<td>EASA.IM.R.S.01460</td>
<td>HELI-ONE USA</td>
<td>FAA STC SH8015SW</td>
<td>BELL 230</td>
<td>Installation of EDZ-705 EFIS four tube flight instrument System. (This STC was classified as Basic by the FAA and is approved using the Post Type Validation Principles.)</td>
<td>HD230/HD-00-002-002 rev 09 dated 22 December 1992 or later approved revision – Drawing List HD2-90-91 dated 22 February 1993 or later approved revision – Flight Manual Supplement.</td>
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<td>EASA.IM.R.S.01462</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA STC SR00401DE</td>
<td>BELL 206L1, 3L4, 4 and 407</td>
<td>Snow deflector installation.</td>
<td>DL-206/407SD-200 revision C Dated 8 January 2007 or later approved revision – Drawing List/PR-206/407SD-120M revision S dated 12 December 2005 or later approved revision – Instructions for Continued Airworthiness</td>
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<td>18/07/2008</td>
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<td>EASA.IM.R.S.01464</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02439NY-D</td>
<td>SIKORSKY S-92A</td>
<td>Installation of a Rockwell Collins SAR-126 Direction Finder System. This STC has been classified as “Basic” by the FAA and has been approved under the PTVP.</td>
<td>FM01366 revision – dated 4 January 2008 or later approved revision – Flight Manual Supplement/D01366 revision B dated 30 September 2007 or later approved revision – Instructions for Continued Airworthiness/DL-01366 revision C dated 26 November 2007 or later approved revision – Installation Drawing.</td>
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<td>10/07/2008</td>
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<td>EASA.IM.R.S.01465</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td>FAA STC SR00334DE</td>
<td>BELL 407</td>
<td>Pneumatic Door Opener (PDO). This STC has been classified as “Basic” by the FAA and has been approved under the PTVP.</td>
<td>DL-407PDO-100 revision ( - ) dated 22 October 1997 or later approved revision – Master Drawing List/PR-407PDO-120M revision 1 dated 12 December 2005 or later approved revision – Instructions for Continued Airworthiness.</td>
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<td>09/06/2008</td>
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<td>EASA.IM.R.S.01466</td>
<td>HELI-ONE USA</td>
<td>FAA STC SH8886SW REV2</td>
<td>BELL 230</td>
<td>Installation of SPZ-700 DAFCS System.</td>
<td>HD230/00-0001 revision R dated 22 December 1992 or later approved revision – Heli-Dyne Drawing List/HD-50-24 dated 08 February 1993 or later approved revision – Flight Manual Supplement.</td>
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<td>EASA.IM.R.S.01467</td>
<td>BELL HELICOPTER Textron, Inc.</td>
<td>FAA SH8421SW</td>
<td>206B, 206-L-3, 206-L-4</td>
<td>Steerable Nose Landing Light. This STC is classified as “Basic” by the FAA and is approved under the Post Type Validation Principles (PTVP).</td>
<td>100-12005 revision N dated 11 February 2006 or later approved revision – Drawing List/AA-12005 no revision dated 13 February 2004 or later approved revision – Instructions for Continued Airworthiness</td>
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<td>09/07/2008</td>
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<td>EASA.IM.R.S.01468</td>
<td>AIR COMM CORPORATION L.L.C</td>
<td>FAA STD SHS28NM</td>
<td>222, 222B, 222J, 230, 430</td>
<td>Installation of a bleed air cabin heating system.</td>
<td>Certification basis</td>
<td>FAA STC 10017042</td>
<td>Pre-installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 1</td>
<td>02/02/2008</td>
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<td>EASA.IM.R.S.01469</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>FAA STC SR009106E</td>
<td>EUROCOPTER DEUTSCHLAND BO-105 EUROCOPTER CANADA BO-105 LS A</td>
<td>Pushout Window Assembly</td>
<td>Installation Data List EDR 01-1458 Rev. B, dated October 30, 2007 or later approved revisions</td>
<td>Installation Instructions for Continued Airworthiness ICA-D105-674, dated 17th March 2000 or later approved revisions</td>
<td>07/01/2009</td>
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<td>EASA.IM.R.S.01471</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SHS316SO</td>
<td>212, 412, 412EP</td>
<td>Pushout Window Assembly</td>
<td>Installation Data List EDR 01-1458 Rev. B, dated October 30, 2007 or later approved revisions</td>
<td>Installation Instructions for Continued Airworthiness ICA-D105-674, dated 17th March 2000 or later approved revisions</td>
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<td>EASA.IM.R.S.01472</td>
<td>HERITAGE AVIATION LIMITED</td>
<td>FAA STC SR03463RC</td>
<td>MD HELICOPTERS 600N MD HELICOPTERS 500N MD HELICOPTERS 369FF MD HELICOPTERS 369E</td>
<td>Heritage Aviation, Ltd. STC SR03463RC; Replacement Instrument Panel (including pedestal, shrouds and glare shield) on applicable MDHI 369E, 369FF, 500N &amp; 600N helicopters.</td>
<td>- Compliance Checklist; E07-1242 Rev A or subsequent approved revision - Master Drawing List; A00-13001 Rev C or subsequent approved revision - Instructions for Cont.A/w; A00-13001 Rev D or subsequent approved revision - Installation Instructions for Cont.A/w; A00-13001 Rev E or subsequent approved revision - Installation Instructions for Cont.A/w; A00-13001 Rev F or subsequent approved revision - Installation Instructions for Cont.A/w; A00-13001 Rev G or subsequent approved revision - Installation Instructions for Cont.A/w; A00-13001 Rev H or subsequent approved revision - Compliance Checklist; A00-13001 Rev A or subsequent approved revision - Master Drawing List; A00-13001 Rev B or subsequent approved revision - Installation Instructions for Cont.A/w; A00-13001 Rev C or subsequent approved revision - Installation Instructions for Cont.A/w; A00-13001 Rev D or subsequent approved revision</td>
<td>1. Applicable to MDHI 369E, 369FF, 500N and 600N helicopters. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the...</td>
<td>19/03/2009</td>
<td>Active</td>
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<td>EASA.IM.R.S.01473</td>
<td>HERITAGE AVIATION LIMITED</td>
<td>FAA STC SR03464RC</td>
<td>MD HELICOPTERS 500N MD HELICOPTERS 369FF MD HELICOPTERS 369E</td>
<td>Heritage Aviation, Ltd. STC SR03464RC; Chelton EFIS installation on applicable MDHI 369E, 369FF &amp; 500N helicopters.</td>
<td>- Compliance Checklist; E07-1242 Rev A or subsequent approved revision - Master Drawing List; A00-13001 Rev C or subsequent approved revision - Instructions for Cont.A/w; A00-13001 Rev D or subsequent approved revision - Compliance Checklist; A00-13001 Rev A or subsequent approved revision - Master Drawing List; A00-13001 Rev B or subsequent approved revision - Installation Instructions for Cont.A/w; A00-13001 Rev C or subsequent approved revision - Installation Instructions for Cont.A/w; A00-13001 Rev D or subsequent approved revision</td>
<td>1. Applicable to MDHI 369E, 369FF, 500N helicopters. 2. Operation with the Chelton EFIS is limited to VFR only. 3. See limitations specified in RPM Supplement. 4. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other...</td>
<td>19/03/2009</td>
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<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
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<td>EASA.IM.R.S.01475</td>
<td>BELL HELICOPTER TEXTRON, Inc.</td>
<td>FAA SR09379RC-D</td>
<td>412 EP</td>
<td>Plot heater failure monitor kit (412-081-001), Stand by altitude battery monitor kit (412-083-001), Crew flashlight holder kit (412-066-001).</td>
<td>- SR00848SE “Optional Accessories Kit” - RFM Supplements - AA-07147, dated 14.04.2008.</td>
<td>1. Applicable to Registration Marks I-OTEL2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.3. d...</td>
<td>05/02/2009</td>
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<td>EASA.IM.R.S.01476</td>
<td>VIH AEROSPACE Inc.</td>
<td>TCCA STC SH08-29</td>
<td>S92A</td>
<td>HS Crew Bubble Window Installation</td>
<td>- VS92-083-MLA, revision 0 dated 16 May 2008 or later EASA approval revision - Master Drawing List: VS92-083-MLA.</td>
<td>Prior to installation of this modification the installer must determine whether the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. This Certificate shall remain valid unless otherwise amended...</td>
<td>25/02/2009</td>
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<td>EASA.IM.R.S.01477</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>FAA STC SR00914SE</td>
<td>EUROCOPTER FRANCE SA 315B</td>
<td>Installation of Dart Bearpaws P/N D315-670-011 - Validation of FAA STC SR00914SE.</td>
<td>DART Aerospace Drawing List MDS-D315-670, revision B, dated 1-January-2005, or later approved revisions.DART Aerospace Installation Instructions IN-D315-670, revision C, dated 1-April-2005, or later approved revisions.DART Aerospace Maintenance Manual MMX-D315-670, revision B, dated 1-April-2005, or later approved revisions.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by this...</td>
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<td>EASA.IM.R.S.01478</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>FAA STC SR00848SE</td>
<td>EUROCOPTER FRANCE SA 315B</td>
<td>Installation of Dart Aerospace Skidtube P/N D315-668-011 (LH) and -012 (RH). Validation of FAA STC SR00848SE.</td>
<td>DART Aerospace Installation Instructions IN-D315-668, Revision A, dated 20-Sept-1999, or later approved revisions.DART Aerospace Maintenance Manual MMX-D315-668, Revision A, dated 13-Sept-1999, or later approved revisions.DART Aerospace Master Drawing List MDS-D315-668, Revision B, dated 1-Jun-2000.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by this...</td>
<td>06/02/2009</td>
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<td>EASA.IM.R.S.01482</td>
<td>GENEVA AVIATION Inc.</td>
<td>FAA STC SH477NMM</td>
<td>AS 350 B, AS 350 B1, AS 350 B2, AS 350 B3, AS 350 BA, AS 350 D</td>
<td>Installation of replacement transmission cowling panels (a.e. Master Drawing List GA119.</td>
<td>- Master Drawing List GA119 Rev. B dated 01/2006 or subsequent revisions of this document...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by third parties...</td>
<td>17/02/2009</td>
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<td>Description</td>
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<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<td>EASA.IM.R.S.01483</td>
<td>GENEVA AVIATION Inc.</td>
<td>FAA STC SR00191SE</td>
<td>EUROCOPTER FRANCE AS 350</td>
<td>Installation of a single place pilot/copilot seat in accordance with MDL GA156</td>
<td>Master Drawing List GA156 Rev. B dated 23 August, 2007, or later approved revisions/Compliance Checklist CCL-156 Rev -- dated May 8, 2008 or later approved revisions</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.1. The installation of this modification b...</td>
<td>27/08/2008</td>
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<td>10017054</td>
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<td>EASA.IM.R.S.01487</td>
<td>PARAVION TECHNOLOGY, INC.</td>
<td></td>
<td>BELL 206, BELL 407</td>
<td>Vent Control Certification.</td>
<td>206VC-100 revision NC dated 22 November 2006 or later approved revision -- Master Drawing List PR-206VC-12M revision 0 dated 22 September 2006 or later approved revision -- Instructions for Continued Airworthiness.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by third parties...</td>
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<td>DART AEROSPACE Ltd.</td>
<td>REV. 1</td>
<td>TCCA STC SH08-40</td>
<td>BELL 412 EP, BELL 412</td>
<td>Replacement Glassshield Installation.</td>
<td>- IIN-D412-776-Rev. B, INSTALLATION INSTRUCTIONS or later approved revision- MDL- D412-776-Rev. B, MASTER DOCUMENT LIST or later approved revision- ICA-D412-776-Rev. 0, INSTRUCTIONS FOR CONTINUED AIRWORTHINESS or later accepted revision</td>
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<td>PRECISE FLIGHT, Inc.</td>
<td>FAA STC SR01821SE</td>
<td>R22, R22 ALPHA, R22</td>
<td>Installation of a Precise Light HID Aviation Lighting System according to FAA STC SR01821SE.</td>
<td>1. Engineering Drawing List (Manufacturing) 020H0000 Rev. C2. Engineering Drawing List (Installation) 590H0000 Rev. B3. Instructions for Continued Airworthiness 590HMAN0003 Rev. A</td>
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<td>PARAVION TECHNOLOGY, INC.</td>
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<td>FAA STC SR00654DE</td>
<td>BELL 206A, 206B, 206L-1, 206L-3, 206L-4, BELL 407</td>
<td>Rotary Latch Belts Installation.</td>
<td>DL-407SB-120 revision A dated 1 November 2007 or later approved revision – Master Drawing List; PA-407SB-120M revision 1 dated 2 November 2007 or later approved revision – Instructions for Continued Airworthiness</td>
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<td>SIMPLEX MANUFACTURING G CO.</td>
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<td>FAA STC SR00886SE</td>
<td>EUROCOPTER FRANCE AS 355 SERIE</td>
<td>Installation of an agricultural spray system. (This STC has been classified as &quot;BASIC&quot; by the FAA in accordance with FAA Order 8110.52 and has been approved using the PTVP Procedures.)</td>
<td>MDL-210 revision B dated 9 February 2001 or later EASA approved revision – Master Drawing List; 210-102005-001 revision 0 R dated 21 December 2000 or later approved EASA revision – Maintenance Manual 210-102003-002 dated 14 March 2001 or later EASA approved revision – Flight Manual Supplement.</td>
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<td>HERITAGE AVIATION LIMITED</td>
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<td>FAA STC SR00481HC</td>
<td>MD HELICOPTERS 600N MD HELICOPTERS 500N MD HELICOPTERS 360F MD HELICOPTERS 369F MD HELICOPTERS 369F</td>
<td>Heritage Aviation Ltd. STC SR00481RC; Installation of noseshelf on applicable MDH 369F, 369F, 369FF, 500N &amp; 600N helicopters.</td>
<td>Compliance Checklist; E07-1240 Rev. A or subsequent approved/modified Master Drawing List; A05-13002 Rev. B or subsequent approved revision. Instructions for Cont.Air; A07-7302 Rev. C (26.4.07) or subsequent approved revision.</td>
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<td>EASA.IM.R.S.01513</td>
<td>AEROQUEST INTERNATIONAL</td>
<td>TCCA STC SH08-52 ISSUE 1</td>
<td>SA 315 B</td>
<td>Installation of Aerial Survey Modification - Validation of Transport Canada (TCCA) STC No. SH08-52.</td>
<td>JCM Aerodesign Limited Flight Manual Supplement 0630025MFMS, dated 9- Oct.-2008 or later approved revisions.</td>
<td>JCM Aerodesign Limited Maintenance Manual Supplement 0630025MFMS, dated 9- Oct.-2008 or later accepted revisions.</td>
<td></td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.2. The installation of this modification by TH...</td>
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<td>EASA.IM.R.S.01514</td>
<td>SIMPLEX MANUFACTURING CO.</td>
<td>FAA STC SR08905SE</td>
<td>AS 350 D AS 350 B</td>
<td>Model 210 Spray System - Installation of an agricultural spray tanksystem</td>
<td>- Simplex Master Drawing List MDL-210, Revision B, dated February 9,2001, or later EASA approved revision.</td>
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<td></td>
<td>Rotorcraft modified in accordance with this STC must be operated in accordance with EASA approved Rotorcraft Flight Manual (RFMS). Simplex Document N°210-102003-002 / EASA, dated 20, May 2009, or later EASA approved revision.</td>
<td>28/05/2009</td>
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<td>EASA.IM.R.S.01518</td>
<td>DART AEROSPACE Ltd.</td>
<td>TC STC SH08-37 ISSUE 1</td>
<td>BELL 214B-1 BELL 214B</td>
<td>D205-778 Comfort Seat Installation</td>
<td>MDL-D205-778 revision A dated 9 September 2008 or later EASA approved revision.</td>
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<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA.IM.R.S.01519</td>
<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02573NY-D</td>
<td>SIKORSKY S-92A</td>
<td>Installation of Main Rotor Blade Recognition Lights.</td>
<td>- IDL01701 revision – dated 9 October 2008 or later EASA approved revision.</td>
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<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA.IM.R.S.01528</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA STC SH09-08</td>
<td>BELL 205A-1 BELL 205A</td>
<td>Replacement Engine Mount Installation</td>
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<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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# List of EASA Supplemental Type Certificates

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<td>EASA.IM.R.S.01529</td>
<td>DART AEROSPACE Ltd.</td>
<td>1</td>
<td>TCA STC SH08-53</td>
<td>BELL 412 EP BELL 412 BELL 214B-1 BELL 214B BELL 212 BELL 206B BELL 205A-1 BELL 2048 AGUSTA AB-112EP AGUSTA AB-112</td>
<td>Procedure Trim Installations (kit - D412-77S)</td>
<td>ICA 412-77S revision A dated 14 November 2008 or later EASA approved revision – Installation Instructions MDL- D412-77S revision A dated 1 December 2008 or later EASA approved revision – Master Document List ICA- D412-77S revision 0 dated 14 November 2008 or later EASA approved revision – Installation...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>25/02/2009</td>
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<td>EASA.IM.R.S.01530</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>REV. 1</td>
<td>FAA STC SR00325E</td>
<td>5CN, 80SN 369HE, 396HM, 396HS 369D, 369E, 396F, 396FF</td>
<td>Installation of Onboard Systems Model 200-231-00, 200-231-01, 200-231-02 and 200-231-03 Cargo Hook Attachment Point Kit in accordance with Onboard Systems Master Drawing List No. 155-038-09, Rev. 11 dated October 26, 2012.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA.IM.R.S.01531</td>
<td>ONBOARD SYSTEMS INTERNATIONAL</td>
<td>REV. 1</td>
<td>FAA STC SR01943SE</td>
<td>BELL 407</td>
<td>200-328-00 Cargo Hook Fixed Provisions Kit 200-329-00 Cargo Hook Suspension Kit with Pillow Blocks 200-330-00 Cargo Hook Suspension Kit without Pillow Blocks 200-331-00 Pin Load Weigh Kit</td>
<td>Installation of Onboard Systems Model 200-231-00, 200-231-01, 200-231-02 and 200-231-03 Cargo Hook Attachment Point Kit in accordance with Onboard Systems Master Drawing List No. 155-038-09, Rev. 11 dated October 26, 2012.</td>
<td>The Certification Basis (CB) for the original product remains applicable to this certificate/approval.</td>
<td>The requirements for environmental protection and the associated certified noise and/or emissions levels of the original product are unchanged and remain applicable to this certificate/approval.</td>
<td>Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>A9118MKII A109 A/C/E/K2/S, A119</td>
<td>Installation for fixed receivers for ground handling blade folding kit</td>
<td>Installation of fixed receivers for ground handling blade folding kit</td>
<td>Installation of fixed receivers for ground handling blade folding kit</td>
<td>Instruction...</td>
<td>Installations of Onboard Systems Model 200-231-00, 200-231-01, 200-231-02 and 200-231-03 Cargo Hook Attachment Point Kit in accordance with Onboard Systems Master Drawing List No. 155-038-09, Rev. 11 dated October 26, 2012.</td>
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<td>APCAL INDUSTRIES, Inc.</td>
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<td>FAA 206-706-341-111, -117 or -123</td>
<td>Installation of a Cable Cutter Kit.</td>
<td>Installation of a Cable Cutter Kit.</td>
<td>Installation of a Cable Cutter Kit.</td>
<td>Instruction...</td>
<td>For P/N 200-320-00 (Cargo Hook Suspension Kit with Pillow Blocks) – PIN 200-328-00 (Fixed Provisions Kit) or Bell Helicopter P/N 206-706-341-111, -117 or -123 (Auxiliary Equipment Kit) – Cargo Hook Provisions Kit without Pillow Blocks...</td>
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<td>KEYSONE HELICOPTER CORPORATION</td>
<td>FAA STC SR02497NY-D</td>
<td>S-76C</td>
<td>Installation of a Honeywell Checkpoint EVXP Vibration Monitoring System</td>
<td>- Rapid Document List RDL01449 Rev A or later approved revision - Instruction for Continued Airworthiness IC01449 Rev. - Flight Manual Supplement FM0729 Rev.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect to the airworthiness of the product. 2. The installation of this modification by third person...</td>
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<td>AIR COMM CORPORATION LLC</td>
<td>FAA STC SR06946DE</td>
<td>EUROCOPTER FRANCE AS 350</td>
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<td>CERTIFIED TECHNOLOGY, INC</td>
<td>FAA STC SR06406SE</td>
<td>369H, 500N, 600N</td>
<td>Rapid Door Removal System Hinges.</td>
<td>- Certification Plan and Regulation Checklist MASCO-001, rev. NC, dated April 16, 2009 - Drawing List CT9602, rev. H, dated April 30, 2009 - Installation Instructions CT9602-1, rev. E, dated January 9, 2009 - Instructions for Continued Airworthiness CT9602-2, rev. -, dated April 16, 2009.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect to the airworthiness of the product. 2. The installation of this modification by third person...</td>
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<td>AAR COMPOSITES, INC.</td>
<td>FAA STC NO. SR03077 AT</td>
<td>S-92A</td>
<td>S-92A Structural Provisions for a Composite Passenger Cabin/IF Interior.</td>
<td>- S92-620-011 revision L dated 22 February 2008 or later approved revision – Master Document List AAR-ER-1116 rev. E dated 25 January 2008 or later approved revision – Instructions for Continued Airworthiness AAR-ER-1204 revision IR dated 21 February 2008 or later approved revision –...</td>
<td>1. The installer is responsible for the addition of decorative finish to continue to comply with JAR 29.853. AAR bulkheads are for interior partitions only and are not intended means of compliance for cargo or smoke retention that the installer must ensure is covered by other design approaches. The...</td>
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<td>EASA.IM.R.S.01541</td>
<td>DART AEROSPACE Ltd.</td>
<td>TCCA SH 08-60</td>
<td>407</td>
<td>206, L-1, L-3, L-4, 208, A-1, B-1, D407-781 Interior Trim Panel Installations.</td>
<td>- IIN-D407-781 revision A dated 22 October 2008 or later approved revision – Installation Instructions; - MDL-D407-781 revision A dated 19 December 2008 or later approved revision – Master Document List; - ICA-D407-781 revision 0 dated 20 November 2008 or later approved revision – Instructions for Continued...</td>
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<td>AIR COMM CORPORATION LLC</td>
<td>FAA.SR00066 DE</td>
<td>BELL 412</td>
<td>Bell Helicopter 412 air Conditioning System.</td>
<td>- DL-412AC Revision E dated 31 July 1997 or later approved revision – Master Drawing List; 412AC-100 dated 10 July 1997 - Flight manual supplement is required for air Comm Corporation drawing 412AC-100 configuration (no temperature control); - 412AC-104 dated 10 July 1997 or later approved revision...</td>
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<td>SIKORSKY GLOBAL HELICOPTERS Inc.</td>
<td>FAA STC SR02283NY-D</td>
<td>SIKORSKY S-92A</td>
<td>Installation of lavatory.</td>
<td>IDO00846C revision A dated 19 December 2005 or later approved revisions – Installation Drawing List – Configuration 1: IDO00846C revision A dated 19 May 2006 or later approved revisions – Installation Drawing List – Configuration 2: IDO00846C revision – dated 31 October 2005 or later appr...</td>
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<td>EASA.IM.R.S.01551</td>
<td>DART AEROSPACE USA, Inc.</td>
<td>FAA STC SR01043SE</td>
<td>AGUSTA A119</td>
<td>Cargo Mirror Installation Note: This STC superseded ENAC STC N-CS-014 being the validation of FAA STC SR01043SE dated 25 January 2003.</td>
<td>FAA STC SR01043SE</td>
<td>- IN-D119-696 Rev. A Installation Instructions or later approved revision - MDL-D119-696 Rev. C Master Document List or later approved revision - ICA-D119-696 Rev. O Instructions for Continued Airworthiness or later approved revision - FMS-D119-696 Rev. A Rotorcraft FM Supplement or later...</td>
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<td>EASA.R.S.00247</td>
<td>AEROSPACE DESIGN FACILITIES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
<td>AS 350 B</td>
<td>Installation of an utility boom mast in accordance with Document ListADF 2003-1 dated 17 December 2003 or later EASA approved revision.</td>
<td>None</td>
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<td>EASA.R.S.00465</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>REV. 2</td>
<td>205A-1</td>
<td>Installation of Air Ambulance Equipment</td>
<td>AR04/951-Component-, Installation-, Operation-, Manual, Illustrated Part Catalog Bell 205, AB 205 Air Ambulance Equipment PIN 205-25-20-100&quot;One Patient Configuration&quot; CMX-BELL205-AA TT&quot;Two Patient Configuration&quot; CMX-BELL205-AATT82. Flight Manual Bell 205 series AB 205 series Supplement, Air Ambulance...</td>
<td>1. Prior to installation of this modification the installer must determinethat the interrelationshipsbetween this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installationof this modification hybrid persons is sub...</td>
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<td>EASA.R.S.00491</td>
<td>MOTORFLUG BADEN-BADEN GmbH</td>
<td>430</td>
<td>Installation of Weather Radar RDR 2000</td>
<td>Flight Manual Supplement Doc. No. Bell 430-01-0681 Revision 0 or later Revision 0 or later EASA approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibilit...</td>
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<td>AEROTECHNIC GMBH</td>
<td>BELL 430</td>
<td>Installation of Cockpit Voice Recorder Universal CVR-120</td>
<td>- Bell 430-03-0101 or later EASA approved Revision / - Flight Manual Supplement Doc. No. Bell 430-03-0581 Revision 0 or later EASA approved Revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or otherminated. 3. This STCa...</td>
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<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td>MD900, MD902</td>
<td>Installation of Air Ambulance Equipment.</td>
<td>AR 015/03-1. Component-, Installation-, Operation-, Manual, Illustrated Part Catalog MD 900 Air Ambulance Equipment PIN 900-25-24-32.000 CMX-MD900-AA74.2. Flight Manual MD 900 Supplement, Air Ambulance Equipment PIN900-25-24-32.000 FMSAT35</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or otherminated. 2. On 20/01/2005</td>
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<td>SCANDINAVIAN AVIONICS DESIGN GMBH</td>
<td>EASA.R.S.00584</td>
<td>AS350 B2/F3/F4/FN</td>
<td>Installation of Bendix/King KX94-405B Radar Altimeter</td>
<td>EASA-approved ECM Supplement 2022-177-AM/01, issue 1 or later EASA approved revision</td>
<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer.</td>
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<td>BUCHER LEICHTBAU AG</td>
<td>EASA.R.S.00587</td>
<td>EUROCOPTER DEUTSCHLAND EC-135-</td>
<td>Ambulance and rescue equipment P/N ARA-EC135-AC65 in accordance with master document list MDCC-AC65 rev D or later approved revision. Revision 1: Change of Document. ARA-EC135-AC65Revision 2: Inclusion of models P2+ and T2+</td>
<td>AR 14/011. Operation-Manual, Installation, Operation Manual, Illustrated Parts Catalog Air Ambulance Equipment SIKORSKY S-76A &quot;quick conversion&quot; P/N 76-25-20-34.00.008-S76-AAT2. Flight manual supplement, Air Ambulance Equipment PIN108-25-20-35.000. FMS-AAT22</td>
<td>1. No smoking is allowed in the rotorcraft cabin. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraphs “Description” and “Associated Technical Documentation”. Compatibility with other aircraft/engine configurations shall...</td>
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<td>EASA.R.S.00580</td>
<td>S-76A</td>
<td>Installation of Air Ambulance Equipment</td>
<td>AR 14/011. Operation-Manual, Installation, Operation Manual, Illustrated Parts Catalog Air Ambulance Equipment SIKORSKY S-76A &quot;quick conversion&quot; P/N 76-25-20-34.00.008-S76-AAT2. Flight manual supplement, Air Ambulance Equipment PIN108-25-20-35.000. FMS-AAT22</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or...</td>
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<td>AR 16/021. Component-&lt; (&gt;,&lt;)Installation-&lt; (&gt;,&lt;) &gt;)Operation-Manual, Illustrated Parts Catalog A 109 Series CMA-A/AAT41 Air Ambulance Equipment PIN108-25-20-30.002. Flight Manual Supplement, Air Ambulance Equipment PIN108-25-20-35.000. FMS-AAT31</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or...</td>
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<td>EASA.R.S.00583</td>
<td>AS 332 L, AS 332 L1, AS 332 L2</td>
<td>Skywatch SKY-497 Traffic Advisory System</td>
<td>Design Report AHSE-515/51, Master Drawing AHSE-E1/50. Rev A, or later approved revision Flight Manual Supplement AHS-014</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or...</td>
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<td>EASA.R.S.00583</td>
<td>AS 332 L, AS 332 L1, AS 332 L2</td>
<td>Techast CFI-503 ELT</td>
<td>Design Report AHSE-515/51, Master Drawing List AHSE-51500 Revision F, Flight Manual Supplement AHS-014 Service Bulletin AHSE-2596-031, latest approved revision</td>
<td>1. This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph “Description”. Compatibility with other aircraft/engine configurations shall be determined by the installer. This certificate shall remain valid unless otherwise surrendered or...</td>
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<td>ESG ELEKTRONIKS TEM-UND LOGISTIK-GMBH</td>
<td>EASA.R.S.00586</td>
<td>EC635 T2+ EC635 P2+ EC635</td>
<td>FUR Mission Work Station EC635 Optional installation of an operator station in the cabin for the operation of mission equipment. The installation is based on ESG Engineering Order TRPA-2L-EO 004 in accordance with ESG Master Drawing List TRPA-MDL-036-01 Issue 1 or later approved revision.</td>
<td>EC135 Flight Manual Supplement - EC135 Flight Manual Supplement, 18RPA/FMS-025, Issue 1 Revision 2 or later approved revision Installation Instructions and Instructions for Continued Airworthiness - Installation Instruction Order, Doc. No. 18RPA IO-024, Issue 1 or later approved revision.</td>
<td>This modification the installation must determine the interaction between this modification and any previous installation modification and prior to installation of this modification the installer must follow the airworthiness of the product. 2. The installation of this modification by third person...</td>
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<td>EUROCOPTER FRANCE EC 120 B</td>
<td>Upgraded Avionics configuration</td>
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<td>EASA.R.S.01020</td>
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<td>AS 332 L2</td>
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<td>EASA.R.S.01022</td>
<td>AERODATA AG</td>
<td>REV. 1</td>
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<td>MBB-BK117 C</td>
<td>Cockpit Arrangement for ADAC BK 117Fleet for IFR operation - Re-arrangement of Cockpit Instruments for a harmonized Cockpit Layout of several BK 117 Helicopters, Furthermore integration of a dual Garmin GNS430 equipment and GTX330 Mode S Transponder</td>
<td>1</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
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<td>EASA.R.S.01023</td>
<td>ECM AVIATION SYSTEMS GMBH</td>
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<td>EUROCOPTER FRANCE AS 360 D</td>
<td>Installation of External Load Box ELB01</td>
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<td>This STC is approved only for the product configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot;. Compatibility with other aircraft engineconfigurations shall be determined by the installer.</td>
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<td>EASA.R.S.01073</td>
<td>AERLYPER S.A.</td>
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<td>EC135 P2</td>
<td>Microwave Transmission System on EC-135 Helicopters</td>
<td>In accordance with Doc No 100198: RFS Supplement Ed. 3</td>
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<td>BO105 C, BO105 S</td>
<td>ASTRY System installation on BO-105 C/ S Helicopters</td>
<td>In accordance with Doc No 100137: RFS Supplement Ed. 4</td>
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<td>EC135 P2</td>
<td>Installation of Image Recorder System UM III and Options System</td>
<td>Doc No 100059: RFS Supplement Ed. 3</td>
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<td>MOTORFLUG BADEN BADEN GmbH</td>
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<td>MD600</td>
<td>Installation of a Standby Altitude Indicator and a KANNAD 406 ELT in a MD600</td>
<td>Installation is restricted to &quot;VFR only&quot;</td>
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<td>AS 355 F, AS 365 N</td>
<td>Instalation of a Gyron 955 Camera mounted on the centerline of the aircraft in front of the nose</td>
<td>Tenencia modification report: TEN 428</td>
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**List of EASA Supplemental Type Certificates**

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### List of EASA Supplemental Type Certificates

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<td>Installation of an L-3 Modular Airborne Data Recording/Acquisition System (MDAS)</td>
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<td>ALIDAUNIA S.R.L.</td>
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<td>AGUSTA A-109 All</td>
<td>MODE No. 010/AVU/2005 - Rotorcraft Flight Manual Supplement: New Emergency Procedures to apply in case of both generators failure.</td>
<td>Supplement No. FMS/AVU/010 Rev. 0 for Type “A” avionics configurationSupplement No. FMS/AVU/010a Rev. 0 for Type “B” and “C” avionics configuration</td>
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<td>Installation of AeroPhone IRIDUM SATCOM/Installation of satellite telephone system AeroPhone on helicopter</td>
<td>Master Document List MDL-IVC-009 Rev. 1 or later:Installation and Testing Instructions REP-IVC-125B Rev. 0 or later:Flight Manual Supplement AFM-070 Rev. 0 or later:Continued Operations Instructions AMM-791 Rev. 0 or later</td>
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<td>Definition and installation BAYO definition file n 16/2005-DEF, revision 0, dated December 19th, 2005, or later EASA approved revision</td>
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<td>SA 365 N S/N 6100</td>
<td>Installation of EMS cabin configuration, in accordance withMBH Technic’s STC file n° 01 MBHT-05, original issue dated 29/03/2005, or later EASA approved revision.</td>
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<td>EASA.R.01215</td>
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<td>New pilot-co-pilot manual quick release pyrotechnic sub-system, PN JD3448AB00100-A0, for approved HEC devices ref. 01-STH-001 and 01-STH-002 and AAN 24527.</td>
<td>Refer to RTE-STH technical file ref. 2005-STH-001&amp;002, ind. 20Definition based on PyroAlliance design file RDB JD3448AB00100 01Installation, Inspection and Maintenance:in accordance with PyroAlliance document STU JD3448AB00100, evolutiona, Rev 10Operation Flight Manuals Supplemtn approved for dr..</td>
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<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>CAA UK RESPONSIBILI TY</td>
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<td>Integration of minor and major changes to Sturm Light, 400MHz ELT, STBY Allimeter, Exhaust heat shield, Autopilot, ADSI.COM/NAV, Transponder, Audiopanel, GPS 530, Compass System and Instrument Panel.</td>
<td>-EASA.R.S.01215</td>
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<td>EASA.R.01225</td>
<td>TATENHILL AVIATION Ltd.</td>
<td>CAA UK RESPONSIBILI TY</td>
<td>AGUSTA A-109 C</td>
<td>Design Data for Compliance with UK CAA Generic requirement 4 by Installation of a battery Low Voltage Indicator.</td>
<td>Tatenhill Aviation Ltd Modification TAL-TAD 025/07TatenhillAviation Ltd RFM supplement TAL-TAD RMS 3.</td>
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<td>EASA.R.01228</td>
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<td>ASC 119-01-00 “Helibasket K3 / Installation of External Basket</td>
<td>Report Helibasket ASC 119-01-00 Part 3, Chapter 4 in accordance with the implementation of the modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect to...</td>
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<td>EASA.R.01231</td>
<td>AERILYPER S.A.</td>
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<td>B0 105 C, B0 105 S</td>
<td>Digital microwave transmission system installation on boardB0-105 C/S.</td>
<td>Doc. No. 100388 “Master Data List”, issue 3, June 2008Doc. No. 100589 “Aircraft Flight Manual Supplement”, issue 1, April 2008Doc. No. 100594 “Continuing Airworthiness Instructions”, issue 00, February 2008.</td>
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<td>Installation of Thales AGILE 2 external camera and associated cabin electronic control unit.</td>
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<td>Installation of the articulated patient loading system into the Bell 427 rotorcraft.</td>
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<td>HELIKOPTER AIR TRANSPORT GmbH</td>
<td>EASA.R.S.01247</td>
<td>EC 135</td>
<td>Installation of Laser Scanning Equipment</td>
<td>EO 85-001-02 Manual Supplement section 9.2-85-001, Rev. 0 or later approved revision Installation for continued airworthinessDoc. No. EO 85-001 chapter 16, Rev. 0 or later approved revision Installation manual Doc. No. IM 85-001, Rev. 0 or later approved revision</td>
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<td>AEROTEC &amp; CONCEPT</td>
<td>EASA.R.S.01248</td>
<td>EC 120 B</td>
<td>Modification des sources d'attitude, d'écart de cap et GPS du système pilote automatique PAB5 SAGEM/Change of sensors sources for attitudes retention, heading error select and GPS steerings of Sagem PAB5 Automatic Pilot system</td>
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<td>STC Holders</td>
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<td>ELECTRONIQUE MAINTENANCE AVIATION</td>
<td>EUROCOPTER FRANCE AS 359H EUROCOPTER FRANCE AS 359F2 EUROCOPTER FRANCE AS 365F1 EUROCOPTER FRANCE AS 365E</td>
<td>Installation of SAFE FLIGHT Power Line Detector pln T101-1, in accordance with EMA STC File no 06/001/AS359H/PLD revision 0, dated January 2006, or later EASA approved revision.</td>
<td>Installation and Continued Airworthiness SAFE FLIGHT installation manual ref. R2025/Operation EMA RFM supplement ref. 06/001/AS359H/PLD revision 0, dated January 2006 or later EASA approved revision.</td>
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<td>AEROSPACE DESIGN FACILITIES Ltd.</td>
<td>CAA UK RESPONSIBILITIES</td>
<td>AS 350 ALL MODELS Installation of Wescam combined nose and side mount</td>
<td>Aerospace Design Facilities Ltd Flight Manual Supplement 9</td>
<td>Approval limited to nose mount only</td>
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<td>MECAR AVIATION GROUP S.p.A.</td>
<td>SAT-COM Installation P/N 6AB1EL018-001The change consists in the installation of a Satellite Telephone System on passenger cabin. This STC supersedes EASA STC EASA R.S. 01254.</td>
<td>The Certification Basis for the original product remains applicable to this certificate/approval. The certificated noise and/or emissions levels of the original product unchanged and remain applicable to this certificate/approval.</td>
<td>None</td>
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<td>CALEDONIAN AIRBORNE SYSTEMS LTD</td>
<td>SIKORSKY S-76 AF-ADELT improved installation on S-76</td>
<td>Flight Manual Supplement 01/02/08/001 Issue ASTC F006/208/001 definition document</td>
<td>STC requires minor change F012/208/001 to be incorporated.</td>
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<td>PREMIAR AIRCRAFT ENGINEERING</td>
<td>CAA UK RESPONSIBILITIES</td>
<td>AS 355 F1 Installation of Honeywell KLNI9 KB GPS</td>
<td>Premier Aircraft Engineering Modification 06-014/Premiar Aircraft Engineering RRF Supplement 06- AS355F-01 Issue A or later approved revision</td>
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<tr>
<td>AEROLITE AG</td>
<td>EC135 Installation of EMS equipment Aerolite PN N 135C70-501</td>
<td>Flight Manual Supplement FMS-06-004, Issue N/C dated 22.05.2006 or later EASA approved revision/Operations Manual OPM-06-008Installation per Drawing List DWSL-06-005</td>
<td>Only S/N 0041</td>
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<td>RTELESEAU DE TRANSPORT D' ELECTRICITE</td>
<td>DGAC-F TN 188 AS 355 N Personnel transport aboard hooked baskets as rotorcraft-load combinations for human external cargo (HEC).</td>
<td>Flight Manual Supplement FMS-06-004, Issue N/C dated 22.05.2006 or later EASA approved revision/Operations Manual OPM-06-008Installation per Drawing List DWSL-06-005</td>
<td>This STC is also applicable on AS355N when equipped with EASA R.S. 01366.</td>
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### List of EASA Supplemental Type Certificates

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<th>EASA STC no.</th>
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<td>EASA.R.S.01274</td>
<td>ECT INDUSTRIES</td>
<td></td>
<td>AS 350 B3</td>
<td>Installation of ECT Industries DMS ARINC data recorder p/n REC-AR0-TT-00</td>
<td>ECT Industries definition file n° OD-0074-XX-82, revision 03, dated 20/10/2006, or later EASA approved revision.Installation ECT Industries installation instructions n° IA-0078-XX-82-FR, revision 00, dated 26/11/2006, or later revision.Instructions for Continued Airworthiness ECT Industries maintenance.</td>
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<td>04/02/2007</td>
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<td>EASA.R.S.01275</td>
<td>AIR MACH S.R.L.</td>
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<td>AS 350 B2</td>
<td>Ki Air Mach ELRAI TV-99-Internal Rack/Installation of electronic equipment for radio bridge</td>
<td>Master Drawing List MDL-INT-001 Rev. 0 Dated 26 April 2009.Instruction for Continued Airworthiness ICA-001 Rev. 0 Dated 26 April 2009.Part List Electronic Equipment PL-TV-99 Rev. 0 Dated 26 April 2006 RPM Supplement S-RFM-001 Rev. 0 Dated 26 April 2006</td>
<td>This installation should not be incorporated in any aircraft unless it is determined that the interrelationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of the aircraft</td>
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<td>10/05/2006</td>
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<tr>
<td>EASA.R.S.01276</td>
<td>AIR MACH S.R.L.</td>
<td></td>
<td>AS 355 F1, AS 355 F2</td>
<td>Ki Air Mach ELRAI TV-99-Internal Rack/Installation of electronic equipment for radio bridge</td>
<td>Master Drawing List MDL-INT-001 Rev. 0 Dated 26 April 2009.Instruction for Continued Airworthiness ICA-001 Rev. 0 Dated 26 April 2009.Part List Electronic Equipment PL-TV-99 Rev. 0 Dated 26 April 2006 RPM Supplement S-RFM-001 Rev. 0 Dated 26 April 2006</td>
<td>This installation should not be incorporated in any aircraft unless it is determined that the interrelationship between this installation and any previously approved configuration will not introduce any adverse effect upon the airworthiness of the aircraft</td>
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<td>EASA.R.S.01277</td>
<td>AEROTEC &amp; CONCEPT</td>
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<td>EC 120 B</td>
<td>Improvement of “VG Fail” alarm for EASA.R.S.01017 and EASA.IM.R.S.01056 installations</td>
<td>Definition and Installation AEROTEC technical file ref. FAM2906.Inspection and Maintenance according to STCs EASA.R.S.01017 and EASA.IM.R.S.01056 directives that remain unchanged. Operation revision 1 dated 24/07/2006 of MAT Aviation’s French Flight Manual Supplement, ref STC Group 5 - No EASA.R.S.01017</td>
<td>Limited to aircraft EC120B S/N 1049, G-VIPR only, when equipped with EASA R.S.01017 and EASA.IM.R.S.01056</td>
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<td>04/06/2006</td>
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<td>EASA.R.S.01295</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
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<td>CAA UK RESPONSIBILITY</td>
<td>MBB-BK</td>
<td>Variable External Payload Configuration</td>
<td>MHL BK 117 C-2 RPM Supplement No. 2 (Weston Mx-10). MHL BK 117 C-2 RPM Supplement No. 3 (NAT Skyhook)</td>
<td>Refer to above RPM Supplements</td>
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<td>EASA STC No.</td>
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<td>EASA.R.S.01306</td>
<td>HELI-ONE (NORWAY) AS</td>
<td>14/12/2006</td>
<td>STC holder</td>
<td>10017192</td>
<td>SA 366 D1, SA 365 NN1, AS 365 NDN3</td>
<td>Installation of Penny &amp; Giles D51615 MPRR CVDR.</td>
<td>- Datasheet - STC Master - One Master Drawing ref. ASHE-52500 revision A, dated 13 January 2007, or later EASA approved revision. - Installation Engineering Order ref. EOM:3130-26597, Revision A dated 21/08/06, or later revision - Instructions for continued airworthiness Maintenance requirements report ref. ASHE...</td>
<td>This STC is approved only for the product configuration as defined inline approved design data referred to in the paragraphs &quot;Description&quot; and &quot;Associated Technical Documentation.&quot; Compatibility with other aircraft configurations shall be determined by the installer. This certificate shall remain...</td>
<td>09/02/2007</td>
<td>Active</td>
<td>1010332</td>
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<tr>
<td>EASA.R.S.01315</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGAL)</td>
<td>11/01/2007</td>
<td>Modification</td>
<td>10017185</td>
<td>BUCHER Rescue Hoist Fatigue Life Extension</td>
<td>H-25-02-06 LIVE TOOLS Annex 2 dated May 2008 or later approved revision – Dossier Vol de Contrôle H-25-02-06 LIVE TOOLS Annex 3 Edition 0 dated 21 May 2008 or later approved revision – Supplément Manuel de VolH-25-02-06 LIVE TOOLS Annex 4 date May 2008 or later approved revision – Plans de Définition...</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product VNE limited to 120 kts. Do not deploy anti...</td>
<td>12/06/2008</td>
<td>Active</td>
<td>1011742</td>
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<td>EASA.R.S.01318</td>
<td>STC TWENTY ONE LIMITED</td>
<td>24/07/2009</td>
<td>STC Twenty One</td>
<td>10017189</td>
<td>STC Twenty One</td>
<td>Installation of rack and sending/reception antennae.</td>
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<td>EASA.R.S.01321</td>
<td>STC TWENTY ONE LIMITED</td>
<td>01/02/2007</td>
<td>CAA UK RESPONSIBILITY</td>
<td>10017188</td>
<td>SIKORSKY S-92N</td>
<td>Description</td>
<td>Heli-One Master Documentation</td>
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<td>EASA.R.S.01326</td>
<td>PHOENIX AEROSPACE LIMITED</td>
<td>09/10/2007</td>
<td>CAA UK RESPONSIBILITY</td>
<td>10017186</td>
<td>EUROCOPTER DEUTSCHLAND BK117 C</td>
<td>Description</td>
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<td>EASA.R.S.01327</td>
<td>ACS FRANCE</td>
<td>16/06/2008</td>
<td>AC350B, AS 350 B1, AS 350 B3, AS 360 BA</td>
<td>10017185</td>
<td>AS 350 B, AS 350 B1, AS 350 B3, AS 360 BA</td>
<td>Installation of Rack and Sending/Reception Antennae</td>
<td>Installation instructions - Technical Order for installation 2029510-TO-01 Iss. 2, dated 27/03/2008.2, Instructions for continued Airworthiness - Continued Airworthiness 2029510-CA-01 Iss. Dated 36/09/2005.</td>
<td>1. The installation of this STC is not approved for Night VFR approved aircraft. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product VNE limited to 120 kts. Do not deploy anti...</td>
<td>11/01/2007</td>
<td>Active</td>
<td>1011741</td>
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<td>EASA.R.S.01328</td>
<td>ACS FRANCE</td>
<td>16/06/2008</td>
<td>CAA UK RESPONSIBILITY</td>
<td>10017184</td>
<td>EUROCOPTER FRANCE EC 120 B</td>
<td>Description</td>
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<td>BUCHER LEICHTBAU AG</td>
<td>24/07/2009</td>
<td>CAA UK RESPONSIBILITY</td>
<td>10017183</td>
<td>REV. 1</td>
<td>Description</td>
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<td>EASA.R.S.01330</td>
<td>AEROTECH &amp; CONCEPT</td>
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<td></td>
<td>SA 365 N</td>
<td>AEROTECH avionic modification n° 6305: installation of two NAV/COM/GPS; GARMIN GNS 430 x 2 for VFR/B/RNAV “En route” operations, with their associated GARMIN GA 96 GPS antenna; two MID CONTINENT MD 41 Annunciation Control Units an ALLIED SIGNAL KA 33 cooling blower, a GARMIN GAD 43 RMI adapter; in accordance with...</td>
<td>Installation of AEROTEC STC Implementation File n° '6305, revision 1 dated 26/10/05, or later revision. Operation AEROTEC EC Flight Manual Supplement n°'6305-AM2, edition 1 revision 0, dated January 29th 2007 or later approved revision. Inspection and maintenance. Refer to individual equipment Manuals.</td>
<td>This STC is limited to SA 365 n° s/n 6218 and can be installed only by AEROTECH technical staff in accordance with AEROTEC STC Implementation File n°'6305, revision 1 dated 26/10/05, or later revision. AEROTEC minor modification n°'6206, approved under EASA R.C.01981, has to be installed as a provision...</td>
<td>12/02/2007</td>
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<tr>
<td>EASA.R.S.01347</td>
<td>HELI-ONE (NORWAY) AS</td>
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<td>SIKORSKY S-61N</td>
<td>AHSE-53880 Breeze 29/30 Stand by Rescue hotel installation.</td>
<td>- RFMS AH5-077 dated 26.08.2008 or later approved revision - Master Drawing AHSE 53880 28.08.2008 or later approved revision - AHSE-53880 Supplement to Rotorcraft Flight Manual (RFM) Aeryper Doc No. 100973 Ed. 01;</td>
<td>1. Prior to installation of this modification the installation must determine that the interrelationship between this modification and any other previously installed modification/® introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>24/06/2008</td>
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<td>EASA.R.S.01348</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
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<td>CAA UK RESPONSIBIL ITY</td>
<td>EC135 T2</td>
<td>Installation of Honeywell Avionics MKX02 EGPWS</td>
<td>Design Report MCH/V 135858 Rev BAFM Supplement No.34 to EC135T2 APM Rev B</td>
<td>As defined in the APM Supplement No. 34</td>
<td>03/04/2007</td>
<td>Active</td>
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<td>EASA.R.S.01350</td>
<td>ECMS AVATION SYSTEMS GMBH</td>
<td>REV. 1</td>
<td>EASA STC 1025</td>
<td>EUROCOPTER DEUTSCHLAND BK117 C</td>
<td>External Abseil Device EAD01/Provides attachment provisions on rotorcraft for fast roping / rappelling and human external Cargo operations. This STC is a revision of the previous EASA STC No. 1025, dated 24.11.2003. The Technical Changes are follows: - the system is extendable, not fixed- the...</td>
<td>Flight Manual Supplement for Fixed External Abseil Device on EC145 (BK 117 C-2), Revision 3, dated 01.01.2007. Maintenance Manual/Use Guide EAD01/MM, Revision 1 dated 01.01.2007.</td>
<td>1. The scope of this STC is limited to attachment provisions for rappelling / fast roping and HEC operations. It does not include all the necessary equipment like ropes and personal carrying device. It does not establish all the procedures necessary for safe operation. It does not constitute an a...</td>
<td>30/01/2007</td>
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<td>EASA STC no.</td>
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<td>EASA.R.S.01352</td>
<td>B/E AEROSPACE FISCHER GMBH</td>
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<td>427</td>
<td>Installation of floor mounted F+E Medical Seats 230/305 into Bell 427 cabin by use of installation racks attached to forward and aft cabin bulkhead fittings</td>
<td>Flight Manual Supplement D0704-021 revision - dated 17 July 2007 or later approved revision/Maintenance Manual Supplement D0704 - 020 revision 0 dated 30 March 2007 or later approved revision/Installation Instructions are contained in the Maintenance Manual Supplement, list of Compliance Documents D0704</td>
<td>Only those in the above F/M Supplement.</td>
<td>20/07/2007</td>
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<td>EASA.R.S.01353</td>
<td>POLICE AVIATION SERVICES Ltd.</td>
<td></td>
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<td>MIWA Increase with PAS External Stores and Surveillance Equipment</td>
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<td>EASA.R.S.01357</td>
<td>FRANCE AVIATION</td>
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<td>EUROCOPTER DEUTSCHLAND EC-135</td>
<td>Installation of Avridyne Advisory System TAS-620.</td>
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<td>BRISTOW HELICOPTERS Ltd.</td>
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<td>S-92A</td>
<td>Installation of Additional Cargo Compartment Baggage Shell</td>
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<td>EASA.R.S.01359</td>
<td>TENENCIA Ltd.</td>
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<td>EUROCOPTER FRANCE AS 350 SERIE</td>
<td>Gyrros 935 camera installation on the AS 350 series a/c.</td>
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<td>TATENHILL AVIATION Ltd.</td>
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<td>AGUSTA A-109 All</td>
<td>Approval of replacement &quot;TOT&quot; gauges</td>
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<td>EASA.R.S.01361</td>
<td>AEROLITE AG</td>
<td></td>
<td></td>
<td>AB139_AW139</td>
<td>The equipment contains a specific cabin floor layer, 2 medical seats installed in the aft part of the cabin, one or 2 stretchers, medical supply cabinet and various stowage compartments</td>
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Note: Detailed limitations and conditions are listed in the associated technical documentation.
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<tr>
<td>RTEREDEAU DE TRANSPORT D'ÉLECTRICITE</td>
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<td>DGAC-FRANCE STC FAM 98TH0595</td>
<td>AS 355 N</td>
<td>Helicopter equipped with ARRIUS 1A modified: name ARRIUS 1E:DGAC:F approved design change 98TH0595 relates to installation of two modified ARRIUS 1A engines on the AS355N helicopter. This installation allows increased engine performance required to perform maintenance on high voltage overhead lines...</td>
<td>• Manual of Installation ARRIUS 1E:X319 HS 0011. - Supplement to manual de vol: DT-ETUD-CNER/StH:PAAM-07-00315 rev.1 • Additif au Manuel d’entretien Turbomoteur GBF 124: Original DGAC approval FAM 98 STH 0595</td>
<td>• Engine operating limitations as defined in the “Installation Manual” reference X319 HS 0011. - Engine maintenance requirements defined in Manual d’entretien Turbomoteur GBF 124.</td>
<td>23/03/2007</td>
<td>Suspende d</td>
<td>10017211</td>
</tr>
<tr>
<td>STC TWENTY ONE LIMITED</td>
<td></td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>R44 II R44</td>
<td>Installation of removable crop spraying equipment onto Robinson R 44 &amp; R 44 II helicopters.</td>
<td></td>
<td>1. Certified in Restricted Category 2. All limitation of RFM supplement apply3. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>01/06/2007</td>
<td>Active</td>
<td>10017212</td>
</tr>
<tr>
<td>AEROLITE AG</td>
<td></td>
<td>EC635 T1, EC635 P2+, EC635 T2+</td>
<td>Armasuisse 2 LH Stretcher Installation. P/ N 13587-501.</td>
<td></td>
<td>• Flight Manual Supplement FMS-07-002. - Operational manual OPM-07-003. • Installation Manual IM-07-001.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>03/07/2007</td>
<td>Active</td>
<td>10017214</td>
</tr>
<tr>
<td>AEROLITE AG</td>
<td></td>
<td>EC635 T1, EC635 P2+, EC635 T2+</td>
<td>Armasuisse 2 RH Stretcher Installation P/ N 13587-501.</td>
<td></td>
<td>• Flight Manual Supplement FMS-07-003. - Operational Manual OPM-07-004. • Installation Manual IM-07-002.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>03/07/2007</td>
<td>Active</td>
<td>10017215</td>
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<tr>
<td>AEROLITE AG</td>
<td></td>
<td>EUROCOPTER DEUTSCHLAND EC-635 EC135 AND EC635</td>
<td>Life Support Module installation P/N 13588-501. The installation consists of a storage provision for up to three oxygen bottles. This is module installed in the forward part of the passenger cabins.</td>
<td>Flight manual supplement: FMS-07-004:Instruction for continued airworthiness and operations manual: OPM-07-005.</td>
<td></td>
<td>31/10/2007</td>
<td>Active</td>
<td>10017216</td>
<td></td>
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<tr>
<td>BOGGI S.R.L.</td>
<td></td>
<td>AS 365 N3</td>
<td>KIT MD365-P001-60. Additional seat fitting for Eurocopter AS365N3 Helicopter in items configuration.</td>
<td></td>
<td>Limited to AS 365N3 SN 6509 and 6386, previously modified by the kit Bucher P/N ARRA-AS365N2-AC50.</td>
<td>04/12/2007</td>
<td>Active</td>
<td>10017219</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.R.S.01376</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td></td>
<td></td>
<td>AS 365 N3</td>
<td>Re-installment of approval for Category A Operations reference FAR 29 Amendment 16</td>
<td>EASA STC 2005-UK.UKCAA AAN28812 McAlpine Helicopters Change Reference MCA/1 365/728/McAlpine Helicopters Project 0223 AS365N3 Serial Number 6648/McAlpine Helicopters RFM Supplement No.6 Honeywell CDS-R Cockpit Display System</td>
<td>As per AFM Supplement TAL-TAD FMS 18 Revision A.</td>
<td>13/04/2007</td>
<td>Active</td>
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<td>EASA.R.S.01377</td>
<td>TATENHILL AVIATION Ltd.</td>
<td></td>
<td></td>
<td>AGUSTA A-109</td>
<td>Design data compliance with Generic Requirement No. 4.</td>
<td>Tatenhill Aviation Master Document List TAL-TAD 015207 Tatenhill Aviation AFM Supplement TAL-TAD FMS 18 Revision A.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>26/07/2007</td>
<td>Active</td>
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<tr>
<td>EASA.R.S.01381</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERING CO RPORATION AG (AEC)</td>
<td></td>
<td></td>
<td>MBB-BK 117 B-2</td>
<td>Avionics Retrofit. Out: ADF System King KDF-806; Dual VHF-COM System Becker AR-3202; VHF-NAV System Becker NR-3300-3; Dual VHF-COM/NAV System King KK-165; Marker System King KR-21; ADF System King KR-87; Transponder Mode S King KT-73 relocation and Autopilot Control Panel...</td>
<td>- Master Data List MDL Doc. No. 662-049; Rotorcraft Flight Manual Supplement Doc. 662-071 Rev. 09 or later approved revisions.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is...</td>
<td>12/11/2007</td>
<td>Active</td>
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<td>EASA.R.S.01393</td>
<td>AIR AMBULANCE TECHNOLOGY GmbH</td>
<td></td>
<td></td>
<td>AS 332 L2</td>
<td>VIP interior quick conversion P/N 332-25-24-3000-501.</td>
<td>1. STC applicable only on individual AS332L2 aircraft equipped with approved VIP salon consisting of Eurocopter furniture: Forward Lounge: 4 face-to-face VIP seats with 2 VIP tables in the middle based on Eurocopter Dwg 332X/81/329 Rear Lounge: 8 comfort seats and 1 purser seat and utilities cabinet...</td>
<td>17/08/2007</td>
<td>Active</td>
<td>1017225</td>
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<tr>
<td>EASA.R.S.01397</td>
<td>PREMIAR AIRCRAFT ENGINEERING</td>
<td></td>
<td></td>
<td>S-76B</td>
<td>Traffic Advisory System Installation.</td>
<td>Premier Aircraft Engineering Limited Modification 07-019 Premier Aircraft Engineering Limited Rotorcraft Flight Manual Supplement No: 07- S76B-04 Issue 1.</td>
<td>1. STC applicable only on individual AS332L2 aircraft equipped with approved VIP salon consisting of Eurocopter furniture: Forward Lounge: 4 face-to-face VIP seats with 2 VIP tables in the middle based on Eurocopter Dwg 332X/81/329 Rear Lounge: 8 comfort seats and 1 purser seat and utilities cabinet...</td>
<td>26/05/2008</td>
<td>Active</td>
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<tr>
<td>EASA.R.S.01400</td>
<td>AEROLITE AG</td>
<td></td>
<td></td>
<td>MBB-BK 117-C2</td>
<td>EMS Stretcher Installation Police Thüringen (Aerolite P/N 145285-501).</td>
<td>Flight Manual Supplement FM-57-009 Rev. N/C, dated 1 September 2008; Operational Manual OPM-07-011</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>12/11/2008</td>
<td>Active</td>
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<tr>
<td>EASA.R.S.01411</td>
<td>PREMIAR AIRCRAFT ENGINEERING</td>
<td></td>
<td></td>
<td>AS 355 N</td>
<td>Installation of a Honeywell KMD850 Multifunction Display replacing the existing IN12A Radar Indicator plus GC360 Radar/Graphics Computer and Interfacing with an existing Honeywell RDR2000.</td>
<td>Engineering Modification 07-352-FMS-07- AS355N-01.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>05/11/2007</td>
<td>Active</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.R.S.01412</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td></td>
<td></td>
<td>EUROCOPTER FRANCE EC 120 B</td>
<td>Installation of Avionics System Garmin GNS 480 / Honeywell KY-96A / Garmin GTX 33 as defined in Engineering Order 2042580.EO</td>
<td>- Definition and Installation in accordance with - Scandinavian Avionics Design Engineering Order 2042580 - EO, issue 1, dated 25.01.2006. - Inspection and Maintenance in accordance with - Scandinavian Avionics Design Instruction for Continued Airworthiness 2042580-CA, issue 1, dated 08.11.2.</td>
<td></td>
<td>19/03/2008</td>
<td>Active</td>
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<tr>
<td>EASA.R.S.01423</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td></td>
<td></td>
<td>EUROCOPTER FRANCE AS 350</td>
<td>Modification H-25-04-07-SFP - installation of deployable antenna mounted on landing gear.</td>
<td>Definition, installation, and inspection of the installation.</td>
<td>- Dossier de Modification - Janvier 2008, Annexe 1 - Dossier Justificatif - Janvier 2008, Annexe 2 - Plans de Definition et P.</td>
<td>This STC provides approval for the installation and deployment of the antenna. A separate approval is required for the installation and operation of any equipment transmitting through the antenna. The installation of this modification by third persons is subject to written permission of the approv...</td>
<td>25/09/2008</td>
</tr>
<tr>
<td>EASA.R.S.01424</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td></td>
<td></td>
<td>EUROCOPTER FRANCE AS 355 SERIE</td>
<td>Modification H-25-04-07-SFP - installation of deployable antenna mounted on landing gear.</td>
<td>Definition, installation, and inspection of the installation.</td>
<td>- Dossier de Modification - Janvier 2008, Annexe 1 - Dossier Justificatif - Janvier 2008, Annexe 2 - Plans de Definition et P.</td>
<td>This STC provides approval for the installation and deployment of the antenna. A separate approval is required for the installation and operation of any equipment transmitting through the antenna. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>25/09/2008</td>
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<tr>
<td>EASA.R.S.01436</td>
<td>ALIDAUNIA S.R.L.</td>
<td>MBB-BK 117 C-1</td>
<td>MB 018AU/2008 *Installation of GPS/ Satellite System usable to locate the helicopter position.</td>
<td></td>
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<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>18/07/2008</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.R.S.01441</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERINGCO RPORATION AG (AIEC)</td>
<td>A 100 K2</td>
<td></td>
<td>Replacement Euronav III by Euronav V (Euroavionics)by removing- the existing RN 5 Euronav III Map Generatorand installing- a new RN 6 Euronav V Digital Map Generator</td>
<td>- Master Data List Doc. No. 725-049, Rev. IR or latest approved revision-RFM S, Doc. No. 725-071, Rev. IR or latest approved revision</td>
<td></td>
<td></td>
<td>Preclude conditions Dual ONS-530A installation- RN 5 Euronav III Map Generator installed with interface to the ONS-530A 2 CLN 14.3 0 Control Unit installed Dynav64.2.0 Display installed ECMS AVIATION AS 355 Series RFM Supplement PADS/BK117/003</td>
<td>28/04/2008</td>
</tr>
<tr>
<td>EASA.R.S.01443</td>
<td>AVIONES, PIEZAS Y ACCESORIOS, S.L.</td>
<td>AS 355 ALL MODELS</td>
<td>HD 1 Ponti Radio Digital video, audio &amp; data Transmission System.</td>
<td></td>
<td>0108002 revision 0 or later approved revision – Master Document List/0108002 revision 0 or later approved revision – Design Configuration:SM 0108002 Revision 0 or later approved revision – Flight Manual Supplement:ICA 0108002 Revision 0 or later approved revision – Instructions for Continued Amens</td>
<td></td>
<td></td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>08/09/2008</td>
</tr>
<tr>
<td>EASA.R.S.01445</td>
<td>GAMA ENGINEERING Ltd.</td>
<td>BELL 412</td>
<td></td>
<td>Installation of a dual electronic horizontal situation indicator (EHSI) system.</td>
<td>RFM85879001 issue 1 dated 19 May 2008 or later approved revision – Rotorcraft Flight Manual Supplement:HL MOD/000 Issue 1 or later approved revision – Modification Statement:ICAS970001 Issue 1 or later approved revision – Instructions for Continued Airworthiness.</td>
<td></td>
<td></td>
<td>1. The EHSI system must have a minimum software level of Eight (8) installed in the symbol generators.2. The YELLOW symbol generator “SG” flag or display unit “DU” flag must not be visible prior to departure.3. Autopilot coupled ADIP tracking is not approved.4. Autopilot operation in HDD...</td>
<td>22/05/2008</td>
</tr>
<tr>
<td>EASA.R.S.01449</td>
<td>SCANDINAVIAN AVIONICS DESIGN ApS</td>
<td>EUROCOPTER FRANCE AS 369H4</td>
<td></td>
<td></td>
<td>Flight Manual Supplement EDX02_FMS-EASA. Original Issue. 01.05.2005Installation Instructions: EDX02_INST_EASA;Maintenances Manual &amp; user Guide: EDX02_MIM_UK_EASA.</td>
<td></td>
<td></td>
<td>The scope of this STC is limited to technical provisions for the helicopter. It does not include ropes and personal equipment, training aspects and operation procedures necessary for the fast roping operation. These procedures must be established by the operator and approved by national authorities...</td>
<td>25/06/2009</td>
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<tr>
<td>EASA.R.S.01451</td>
<td>ECMS AVIATION SYSTEMS GMBH</td>
<td>EUROCOPTER FRANCE AS 365C2</td>
<td>External Load Attachment Device EDX02.</td>
<td></td>
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<td>18/07/2008</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.R.S.01456</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>AS 365 N3</td>
<td>Installation of NAT 611/614 Voice Warning Generator (AVAD).</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>EASA R.S.01467</td>
<td>None.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
<td>22/04/2008</td>
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<tr>
<td>EASA.R.S.01464</td>
<td>AERODATA AG</td>
<td>EC135 T2, EC135 T2+</td>
<td>Airborne observation Platform (AOP) Integration in (EC135). Installation of a Bar SafeHD FLIR on the right skid and an operator station in the passenger compartment based on Aerodata change no. EC135-3.02 (see ML/A/C-132 Rev. 1 or later approved revisions.</td>
<td>Flight Manual Supplement AMM-956, Rev. 1, Maintenance Manual Supplement AMM-956, Rev. 0, Installation Instruction HB-A/C-957, Rev. 1.</td>
<td>None.</td>
<td></td>
<td></td>
<td></td>
<td>29/08/2008</td>
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<td>EASA.R.S.01471</td>
<td>KUERZI AVIATION AG</td>
<td>BELL 206B</td>
<td>Radar altimeter installation including adapter plate.</td>
<td>FMS-200716286 original issue dated 17 December 2008 or later EASA approved revision – Flight Manual Supplement, ICA-200716286 revision N/ C dated 28 November 2008 or later/EASA approved revision – Instructions for Continued Airworthiness: MLD-200716286 revision NIC dated 1 December 2008...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>18/12/2008</td>
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<td>EASA.R.S.01474</td>
<td>AEROLITE AG</td>
<td>REV. 1</td>
<td>AB139, AW139</td>
<td>EMS Provisions Kit. Revision of STC for incorporation of PAX seat configurations Aerospat P/N 139117-506, 139117-506 and 139117-501f 12, 14 and 15 PAX seats</td>
<td>FMS-08-02 Revision AMDL-08-005 Revision ADM-08-002 Revision A</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons...</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
<td>Status</td>
<td>System Record</td>
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<td>EASA.R.S.01475</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>EC135 T2+; EC135 PD+</td>
<td>Installation of STAR SAFIRE HD FLIR (based on EC UK modification MHU/135/1116).</td>
<td>- Flight Manual Supplement: EC UK EC135T2+ EC135PD+ FMS No. 44 Issue 1o; If in addition the Retractable Digital Microwave Downlink Antennas, i.e. Eurocopter UK List Change No. MCH/1 A135/791 is installed:EC UK EC135T2+ EC135PD+ FMS No. 42 Issue 1o Installation Instructions, Instructions f...</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third party...</td>
<td>14/11/2008</td>
<td>Active</td>
<td>10017204</td>
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<tr>
<td>EASA.R.S.01481</td>
<td>AEROLITE AG</td>
<td>EC135</td>
<td>EC135 EMS Equipment Update of (R) FMS-401 Rev.B or later (R)FMS-08-006 Rev.A or laterOPM-08-006 Rev. A or laterMDL-401 Rev.B or later</td>
<td>Helicopter shall be operated as per limitations section of FMS Placards as per limitations section of FMS shall be installed. No smoking permitted when Oxygen is installed. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>24/06/2008</td>
<td>Active</td>
<td>10017299</td>
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<tr>
<td>EASA.R.S.01488</td>
<td>AVIONICS INTEGRATION &amp; ENGINEERINGCOMPUTATION AG (AIEC)</td>
<td>EC-135-T2</td>
<td>COM/NAV/ATC Avionics System Installation.</td>
<td>AIEC Master Data List E0743-049, Rev. B or later approved Revision.</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>04/02/2009</td>
<td>Active</td>
<td>10017299</td>
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<td>EASA.R.S.01489</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>AS 355 NP</td>
<td>Installation of a Leo II FLIR System</td>
<td>Flight Manual Supplement No 18 – Leo II FLIR System – Issue 1 or later EASA approved revision:Maintenance Manual Supplement MMS/3551127-001 Issue A dated 5 June 2008 or later EASA Approved revision:Top Level Installation Drawing A3/3551127-100</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>13/01/2009</td>
<td>Active</td>
<td>10017299</td>
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<td>EASA.R.S.01490</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>AS 355 NP</td>
<td>Installation of SFIM 85T31 3 Axis Autopilot System</td>
<td>Flight Manual Supplement No 29 – SFIM 85T31 (3 Axis) Autopilot VFR – Issue 1 or later EASA approved revision:Maintenance Requirements Document MR/3551130-001 Issue 1 or later EASA approved revision:AS/A3/3551130-100 – GA of 3 Axis Autopilot System – Issue A</td>
<td>0 Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>14/01/2009</td>
<td>Active</td>
<td>10017279</td>
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<tr>
<td>EASA.R.S.01491</td>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>AS 355 NP</td>
<td>Installation of an alternate Pitot System</td>
<td>Eurocopter UK Ltd. AS/355NP Flight Manual Supplement No 28 dated 3 November 2008 or later EASA approved revision:Eurocopter UK Ltd. AS/355NP drawing A3/3551130-T01.1 – GA of Alternate Pitot System – Issue AE/Eurocopter UK Ltd. AS/355NP report MR/3551133-001 Issue B dated 22 October 2008 or later EASA approved revision:</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>11/12/2008</td>
<td>Active</td>
<td>10017271</td>
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<td>STC Holders</td>
<td>STC No.</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>STC ref.</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of issue</td>
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<tr>
<td>MBB-BK 117-C2</td>
<td>10043035</td>
<td>AERODATA AG</td>
<td>INSTALLATION OF EMS EQUIPMENT</td>
<td>P/N 145268-501 including: 3 medical seats, a primary patient system, medical supply unit, bottle rack, medical stowage cabinet, EMS electrical installation, medical equipment retainers, tie down web and stretcher stowage provisions.</td>
<td>1. Flight manual supplement FMS-08-007 issued 1st September 2008 or later approved version.</td>
<td>Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by third persons...</td>
<td>14/11/2008</td>
<td>Active</td>
<td>10017274</td>
</tr>
<tr>
<td>BUCHER LEICHTBAU AG</td>
<td>10017274</td>
<td>AEROSPACE DESIGN FACILITIES Ltd.</td>
<td>INSTALLATION OF VIP-EC135-AOE3 BUCHER VIP EQUIMENT(REVISION OF FOCA STC FOCA Z 25-20-64 ISSUE 3)</td>
<td>EC-135-P1, EC-135-P2,</td>
<td>- (R)FM Supplement FMA-3 Rev. A or later approved revision- OMM-VIP-EC135- AOE3 Revision V/C-Master Document List MDQC- AOE3 Revision D- FOCA STC Data sheet 2 Z 25-20-64 Issue 3or later revisions of the above listed documents approved by EASA</td>
<td>Not applicable</td>
<td>14/11/2008</td>
<td>Active</td>
<td>10017275</td>
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<tr>
<td>AERODATA AG</td>
<td>10033443</td>
<td>Heli-Union S.A.</td>
<td>STC Holder: Optional installation of a GARMIN 530 COM/NAV/GPS</td>
<td>EC-135-P1, EC-135-P2,</td>
<td>- EC135 Flight Manual Supplement, EC135 Flight Manual Supplement,FMS.EU- EC135-34-131 Rev. 4, dated 09.05.2009, or later approved revision. Technical Instructions: Installation and Maintenance Manual, SMMA-TMPA 0120, Rev. 0</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by third persons...</td>
<td>09/06/2008</td>
<td>Orphaned</td>
<td>10017276</td>
</tr>
<tr>
<td>BUCHER LEICHTBAU AG</td>
<td>10017275</td>
<td>AVIATION SUPPORT GmbH</td>
<td>INSTALLATION OF HIGH DEFINITION (HD) BROADCASTING SYSTEM</td>
<td>EC-135-P1, EC-135-P2,</td>
<td>- EC135 Flight Manual Supplement, EC135 Flight Manual Supplement,FMS.EU- EC135-34-131 Rev. 4, dated 09.05.2009, or later approved revision. Technical Instructions: Installation and Maintenance Manual, SMMA-TMPA 0120, Rev. 0</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2. The installation of this modification by third persons...</td>
<td>09/06/2008</td>
<td>Orphaned</td>
<td>10017277</td>
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<td>STC Holder</td>
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<td>Foreign/NAA STC ref</td>
<td>Model</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>CA1 UK RESPONSIBILITY</td>
<td></td>
<td>EC135 T2+; EC135 P2+</td>
<td>FSI STAR SAFIRE FLIR SKU: MOUNTED Optional installation of an external FLIR camera FSI Star Safire on the right hand multifunction carrier. The installation is based on EC UK change MHL/135/1155 and in accordance with drawing number A4/1351155-100-1. Issue A or later approved revision.</td>
<td>1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) The installation of this modification by third party.</td>
<td></td>
<td>09/06/2009</td>
<td>Active</td>
<td>1017231</td>
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<tr>
<td>ECMS AVATION SYSTEMS GMBH</td>
<td></td>
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<td>EUROCOPTER FRANCE AS 350 D EUROCOPTER FRANCE AS 350 BB EUROCOPTER FRANCE AS 350 BA EUROCOPTER FRANCE AS 350 B3 EUROCOPTER AS 350 B2 EUROCOPTER FRANCE AS 350 B1 EUROCOPTER FRANCE AS 350 B EUROCOPTER FRANCE AS 350 B</td>
<td>External Antenna for Transceiver System AT501 installation on the ldg gear crossblade.</td>
<td>Flight Manual Supplement: BK117 B-2 Flight Manual Supplement FMS 10-4, AT501 Rev. A: Installation Instructions / Instructions for Continued Airworthiness</td>
<td>1. This STC provides approval for structural and electrical provisions only. Installation and operation of any equipment to be connected to the external antenna shall be covered by separate approval. 2. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) The installation of this modification by third party.</td>
<td>03/03/2009</td>
<td>Active</td>
<td>1017283</td>
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<tr>
<td>DRF STIFTUNG LUFTRETUNG EEMNKHUTIGE AG</td>
<td></td>
<td></td>
<td>EUROCOPTER DEUTSCHLAND BK117 B</td>
<td>Title: Installation of a L-3 SSCVR Cockpit Voice Recorder Description: Optional Installation of a Cockpit Voice Recorder. The installation is in accordance with DIF change no. 135, MOD. MID-117-31-32-00001.</td>
<td>Flight Manual Supplement: BK117 B-2 Flight Manual Supplement FMS 10-4, AT501 Rev. A: Installation Instructions / Instructions for Continued Airworthiness</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. 2) The installation of this modification by third party.</td>
<td>04/04/2009</td>
<td>Active</td>
<td>1017285</td>
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<td>SAFRAN HELICOPTER ENGINES</td>
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<td>DGAC FRANCE STC NO44- SF-1122</td>
<td>Installation of HESIS thermic camera system manufactured by SAGEM</td>
<td>- Definition: CGTM technical note NT 033/PR/04-20 - 29/06/2002 - Installation: CGTM technical note IT 033/0YC6/-43/01 dated 21.05.2002 and SAGEM Note: HESIS-GEN 51056Doc00/51056SAD - Maintenance: CGTM technical note NT 033/0YC6/-79/01 dated from 13.12.2001. Ope..</td>
<td>1. Only Turbomeca personal or specifically trained mechanics are allowed to install fixed parts and components of the HESIS kit. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>05/01/2009</td>
<td>Active</td>
<td>1017287</td>
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<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>CA1 UK RESPONSIBILITY</td>
<td></td>
<td>MBB BK117 C-2</td>
<td>- Installation of a MX 15 FLIR, Skid Mounted (Meeker Mount)- Installation of a MX 15 FLIR in combination with the installation of EASA.IM.R.S.01449. The installation is based on EC UK MHU 145/1157 (a.w. Drw. No. A2/1451157-100.</td>
<td>- Aircraft Flight Manual Supplement: Supplement No. 05 &quot;WESCAM MX15 FLIR Surveillance System&quot;, issue No. 1 or later approved revision- Instructions for Continued Airworthiness and Installation Instructions- Maintenance Requirements Document, MRL-1451157-001, Issue A or later approved...</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAA STC ref.</td>
<td>Models</td>
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<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<td>EASA.R.S.01539</td>
<td>AIRLIFT A/S</td>
<td>AS 365 N3</td>
<td>EMS kit installation</td>
<td>-Installation in accordance with Airlift AS Engineering Order E0-365-25-103 Rev 0 or later approved revisions. -Airlift AS Instruction for Continued Airworthiness ICA - 365-25-103 Rev 1 or later approved revisions. -Airlift AS Flight Manual Supplement FMS-365-25-103 Rev. 2 or later approved revisions...</td>
<td>1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the product.2) The installation of this modification by third party...</td>
<td>27/05/2009</td>
<td>Active</td>
<td>10017202</td>
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<td>EASA.R.S.01556</td>
<td>AEROLITE AG</td>
<td>AB412</td>
<td>Quick change EMS Kit</td>
<td>- (R)FMS-09-003 Rev. NC-OPM-09-005 Initial Issue, dated March 19, 2009- MOL-09-004 Rev. N/C- EQL-09-004 Rev. NC</td>
<td>1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the product.2) The installation of this modification by third party...</td>
<td>24/07/2009</td>
<td>Active</td>
<td>10043393</td>
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<td>EASA.R.S.01560</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td>EUROCOPTER FRANCE AS 355 SERIE</td>
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<td>EASA.R.S.01561</td>
<td>SOCIETE D'ETUDE ET DE GESTION AERONAUTIQUE (SEGA)</td>
<td>EUROCOPTER FRANCE AS 350 SERIE</td>
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<tr>
<td>EASA.R.S.01562</td>
<td>ALDAUNIA S.R.L.</td>
<td>A109 S</td>
<td>Sat phone system usable for transmission of voice, data and GPS position.</td>
<td>- BM No. 021/ALI/2008 Rev. 0- SFM FMS/ALI/021/Rev. 0</td>
<td>none</td>
<td>15/05/2009</td>
<td>Active</td>
<td>10043060</td>
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<tr>
<td>EASA.R.S.01563</td>
<td>DRF STIFTUNG LUFTrettung EMEINNUSTIGE AG</td>
<td>BK117 C-1</td>
<td>Title: Installation of Modular Airborne Data Recording/Acquisition System (MADRS) Description: Optional Installation of a Cockpit Voice and Flight Data Recorder System. The installation is in accordance with DRF change no. 111 (=) MOL-MD-117-31-32-00003 Rev. A or later approved revision.</td>
<td>CS-29 Amdt. 2</td>
<td>None</td>
<td>23/06/2009</td>
<td>Active</td>
<td>10034339</td>
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<tr>
<td>EASA.R.S.01566</td>
<td>ECMS AVIATION SYSTEMS GMBH</td>
<td>BELL 206L-4</td>
<td>Installation of Bifilar Cargo System</td>
<td>Flight Manual Supplement Doc. No. BC501_FMS_EASA_R0, dated 09.04.2009</td>
<td>1) Prior to installation of this modification the installer must determine that the interrelationship between this modification and any previously installed modification will introduce no adverse effect upon the airworthiness of the product.2) The installation of this modification by third party...</td>
<td>08/06/2009</td>
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<td>STC Number</td>
<td>Rev. Level</td>
<td>Foreign/NA</td>
<td>STC ref.</td>
<td>Model(s)</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
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<tr>
<td>AIRBUS HELICOPTERS UK LIMITED</td>
<td>EASA.R.01576</td>
<td>A</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>EC135 T2, EC135 P2+, EC135 T2+</td>
<td>EC135 T1, EC135 T1 (CPDS)</td>
<td>Buker 16g Stretch</td>
<td>Flight Manual Supplements - When installed in EC135 T1 (CPDS): McAlpine Helicopters Ltd. Supplement No. 5, Buker 16g Stretch Installation, Issue 1 or later approved revision - When installed in EC135 T2 (CPDS): McAlpine Helicopters Ltd. Supplement No. 12, Buker 16g Stretch Installation</td>
<td>1. Prior to installation of this modification the installer must determine that the interpretation between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product.</td>
<td>15/06/2009</td>
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<tr>
<td>HELI-ONE (NORWAY) AS</td>
<td>EASA.R.01581</td>
<td>A</td>
<td>CAA UK AAN 25867 ISSUE 2</td>
<td>SKORSKY S-61N</td>
<td>Installation of Automatically Deployable Emergency Locator Transmitter (ADELT)</td>
<td>- Bond Modification No. 1731 - Flight Manual Bond Supplement No. 5 or later approved revision: No This STC reflects the transfer of CAA UK AAN 25867 Issue 2 from Bond Helicopters Ltd. to Heli-One Norway AS</td>
<td>Prior to installation of this modification the installer must determine that the interpretation between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. This Certificate shall remain valid unless otherwise shown.</td>
<td>06/07/2009</td>
<td>Active</td>
</tr>
<tr>
<td>EUROCOPTER UK Ltd.</td>
<td>EASA.R.01592</td>
<td>A</td>
<td>CAA UK RESPONSIBIL ITY</td>
<td>EC135T2, EC135P2+, EC135T1, EC135T2</td>
<td>Installation Martin Baker BaerBellinger passenger Seat: Note: This STC was issued to extend CAA UK AAN 27085 issued to McAlpine Helicopters Ltd. to the models EC135 P2+ and EC135 T2+.</td>
<td>Eurocopter UK Supplement 6 issue 1 part of Eurocopter UK change referred MCH/135/496/HUK CAA AAN No 27085 issue 3 provided original approval. EASA STC to be based to include addition of T2+ and P2+ variants.</td>
<td>none</td>
<td>29/06/2009</td>
<td>Active</td>
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<tr>
<td>LUFTHANSA TECHNICK AG</td>
<td>LBA.21E.TA0742/NO.709</td>
<td>A</td>
<td></td>
<td>A321-211</td>
<td>Having regard to the technical visa granted by LBA (Germany) and referenced to as 'STC Nr LBA 21 E2.1 A0742', as described in Annex 2 (A) to the Drying System Installation</td>
<td>Having regard to the technical visa granted by LBA (Germany) and referenced to as &quot;STC Nr LBA 21 E2.1 A0742&quot;, as described in Annex (A) to the Drying System Basis is established at LBA Certification Reviewitem CR1 A-100, Issue 1, dated June 06, 2003.</td>
<td>Having regard to the technical visa granted by LBA (Germany) and referenced to as &quot;STC Nr LBA 21 E2.1 A0742&quot;, as described in Annex (A) to the Drying System Basis is established at LBA Certification Reviewitem CR1 A-100, Issue 1, dated June 06, 2003.</td>
<td>05/11/2003</td>
<td>Active</td>
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<td>THE NORDAM GROUP LLC</td>
<td>LBA.RC1963/EASA.1M.R.0037</td>
<td>A</td>
<td>FAA STC</td>
<td>MD900</td>
<td>Installation of a sliding window in cabin sliding door</td>
<td>Nordam Transparency Division of Texas drawing number 9304951, revision D, dated April 24, 1998 or later EASA approved revision.</td>
<td>1. The Nordam Flight Manual Supplement dated August 24, 1995 is required when operating with the installed sliding window in cabin sliding door 2. This STC is approved only for the configuration as defined in the approved design data referred to in the paragraph &quot;Description&quot; Compatibility with other...</td>
<td>13/07/2004</td>
<td>Active</td>
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<td>EASA STC no.</td>
<td>STC Holder</td>
<td>Rev. level</td>
<td>Foreign/NAAC/STC ref.</td>
<td>Model</td>
<td>Description</td>
<td>Certification basis</td>
<td>Associated Technical Documentation</td>
<td>Limitations/Conditions</td>
<td>Date of Issue</td>
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<tr>
<td>LBA/RC1064 / EASA.IM.R.S.0041</td>
<td>THE NORDAM GROUP LLC</td>
<td>FAA STD/ SR0916SRC</td>
<td>MD900</td>
<td>Installation of a cockpit door sliding window</td>
<td>NORDAM drawing 9004662(1,1) Revision B dated September 2002 or later EASA approved Revisions, Installation Instructions NTDSTC-006, Revision B dated August 28 2002 or later EASA approved Revisions.</td>
<td>NORDAM drawing 9004662(1,1) Revision B dated September 2002 or later EASA approved Revisions, Installation Instructions NTDSTC-006, Revision B dated August 28 2002 or later EASA approved Revisions.</td>
<td>1. Continued Airworthiness must be maintained in accordance with MD Helicopters, Inc. Maintenance Manual CSH-90RMA-2, Revision 17, dated 27/06/2002, or later EASA approved Revision. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person...</td>
<td>13/07/2004</td>
<td>Active</td>
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<tr>
<td>LBA/S1357 (2004-248)</td>
<td>AIR PLANES SERVICES, CORPORATION</td>
<td>FAA STC SA44265W</td>
<td>CESSNA 172R</td>
<td>The installation of 180 horsepower Lycoming 0-360-A3A or IO-360-L2A and propeller McCauley IA170/CFA7690</td>
<td>see Certification Basis of STC SA44265W Drawing List 1728803 Sh 1 &amp; Sh 2 and 1728806 Sh 3, FAA-approved on June 12, 1988 or later FAA approved revision</td>
<td>Drawing List 1728803 Sh 1 &amp; Sh 2 and 1728806 Sh 3, FAA-approved on June 12, 1988 or later FAA approved revision</td>
<td>Cessna Model 172R limited to IO-360-L2A engine only</td>
<td>13/01/2004</td>
<td>Active</td>
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<tr>
<td>P-EASA.IM.A.S.03168</td>
<td>INNOTECH AVIATION DIVISION OF IMP GROUP LTD</td>
<td>BOMBARDIER BD-700-1A10</td>
<td>BOMBARDIER BD-700-1A10</td>
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<td>03/07/2009</td>
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<td>P-EASA.R.S.01559</td>
<td>POLICE AVIATION SERVICES Ltd.</td>
<td>CAA UK RESPONSIBILITY</td>
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