

# Part 21 Light – Overview and Next Steps AERO 2023

20<sup>th</sup> April 2023

Carl GARVIE - Regulations Officer/PCM – Initial Airworthiness CT.5.1 Initial Airworthiness Standards and Specifications

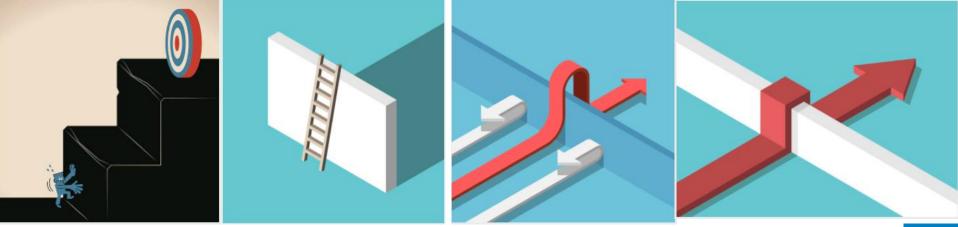


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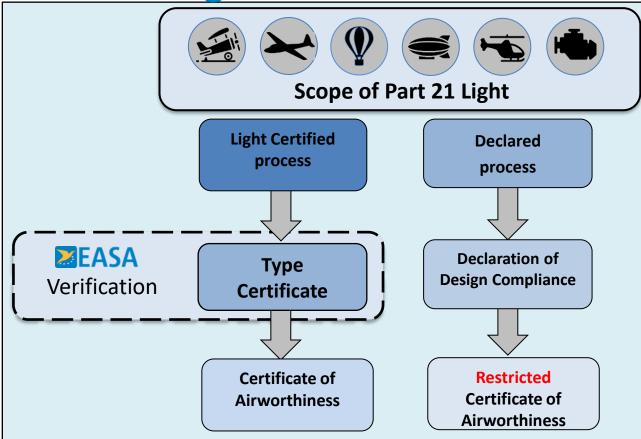
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# Part 21 Light- Main objectives

Enable the application of a **proportionate approach** for products that are considered to **pose less risk** when compared with other more complex products.



## **Part 21 Light- Overview**







# Scope of Part 21 Light

#### **Light Declared aircraft**

#### **Light Certified aircraft**

Aeroplanes with a MTOM of <b>1 200 kg</b> or less with a seating configuration of <b>maximum 2 persons</b> ;		Aeroplanes with a MTOM of <b>2 000 kg</b> or less with a seating configuration of maximum <b>4 persons</b> ;
Sailplanes or powered sailplanes of <b>1 200 kg</b> MTOM or less;	~	Sailplanes or powered sailplanes of <b>2 000 kg</b> or less;
Balloons designed for maximum 4 persons;		All balloons;
Hot air airships designed for maximum 4 persons.		Hot air airship <i>s;</i> Passenger gas airships designed for maximum 4 persons;
Je Son		Rotorcraft with a MTOM of <b>1 200kg</b> or less with a seating configuration of maximum <b>4 persons</b> ; Gyroplanes;
		Piston engines and fixed pitch propeller on above





MFASA

- → An application for a Type Certificate is still required (EASA Forms have been adapted)(211.A.24);
- → To be eligible (21L.A.22) for a Type Certificate applicant must be either:
  - → An approved design organisation (i.e. DOA holder) under Part 21 Subpart J;
  - → A declared design organisation under Part 21 Light Subpart J.
- → For production of a certified product the organisation can be either (21L.A.143 (c)):
  - $\rightarrow$  An approved production organisation (i.e. POA holder) under Part 21 Subpart G;
  - → A declared production organisation under Part 21 Light Subpart G.
- → The certification basis for the product is established and notified to the applicant by EASA (21L.B.43);
- → A compliance demonstration plan is prepared by the applicant and approved by EASA (21L.A.24 (b)(21L.B.46(a));









- Compliance documentation is provided to EASA in case of need (21L.A.25 (a)&(b));
- → <u>Before 1<sup>st</sup> flight</u> and approval of the flight conditions and issue of the permit to fly:
  - $\rightarrow$  EASA conducts a 'critical design review of the design and a physical inspection and assessment of the aircraft' (21L.B.242 (a)(1))
  - $\rightarrow$  NAA conducts a 'physical inspection of the aircraft' (21L.B.241 (a)(1))
- $\rightarrow$  The purpose is for:
  - → EASA: to ensure that the aircraft is capable of safe flight, and that flight testing can be conducted safely;
  - → The NAA: to be satisfied that the aircraft conforms to the design for the requested permit to fly;



 $\rightarrow$  The secondary purpose is for:

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- EASA: to conduct a first oversight visit of the declared design  $\rightarrow$ organisation;
- The NAA: to conduct a first oversight visit of the declared production  $\rightarrow$ organization.
- $\rightarrow$  The location is at the facilities of the applicant and the physical location of the aircraft that will be issued with the Permit to Fly;
- $\rightarrow$  Findings can be raised against the aircraft and the declared design or production organization;
- $\rightarrow$  If there are no findings (or findings have been resolved) EASA approves the flight conditions and the NAA issues the permit to fly.









- → At the end of compliance demonstration all compliance documentation is provided to EASA;
- → The applicant declares compliance with the agreed certification basis (21L.A.25 (f));
- $\rightarrow$  At this stage in the process this then triggers:
  - $\rightarrow$  EASA to conduct 'a physical inspection and assessment of the first article of that product in the final configuration' (21L.B.46 (c)&(d))
  - $\rightarrow$  The NAA to conduct 'a first article inspection' (21L.B.143 (b) and 21L.B.161 (c)).
- $\rightarrow$  Location:

MFASA

→ The location is at the facilities of the applicant and the physical location of the aircraft that will be issued inspected and assessed.









- $\rightarrow$  Purpose of 'first article inspection':
  - $\rightarrow$  For EASA to:
    - → 'Verify the compliance of the product with the applicable type-certification basis and the applicable environmental protection requirements';
    - $\rightarrow$  Conduct oversight of the declared design organization.
  - $\rightarrow$  For the NAA to:
    - → Ensure conformity of the aircraft for the first Certificate of Airworthiness (and Noise Certificate);
    - $\rightarrow$  Conduct oversight of declared production organization;
- $\rightarrow$  Timing;
  - → Should <u>not</u> be considered to be a single one day event and could be a series of visits before and after the declaration of compliance (e.g. noise testing).





- $\rightarrow$  Means of conducting the 'first article inspection':
  - → Gathering evidence to support compliance through a physical inspection and assessment of the aircraft and, if applicable, of the engine and the propeller;
  - $\rightarrow$  Other forms of evidence include:
    - → Witnessing or participating in live testing (including flight testing) of the aircraft, engine, propeller, systems or components;
    - $\rightarrow$  Evaluation of the final compliance-demonstration plan;
    - $\rightarrow$  Evaluation of the completeness of the declaration of compliance;
    - → Evaluation of supporting compliance documentation and test reports;
    - $\rightarrow$  Discussions with key design and production personnel;
    - $\rightarrow$  Review of design processes and procedures.

- $\rightarrow$  If no findings are raised then:
  - $\rightarrow$  EASA issues the Type Certificate (21L.B.47 (a))
  - → NAA issues the first Certificate of Airworthiness and Noise Certificate (21L.B.162 (a) & 21L.B.172 (a))





# **Specifics of Part 21 Light – Declared**



### Specifics of Part 21 Light –Declared

- → An application is not required but the declarant is encouraged to contact EASA as early as possible to initiate a project;
- → There are no formal design organisation requirements for the declarant (although they have obligations for design);
- → For production an organisation can be (211.A.143 (d)):
  - An approved production organisation (i.e. POA holder) under Part 21 Subpart G;
  - → A declared production organisation under Part 21 Light Subpart G.
  - → A production organisation using Part 21 Light Subpart R

FASA

- → The declarant selects the applicable 'technical specifications' that will be the basis for the declaration of design compliance (21L.B.61);
- → A compliance demonstration plan is prepared by the declarant and provided to EASA (but not approved) (211.A.44 (a)).









### Specifics of Part 21 Light –Declared

- → Compliance documentation is provided to EASA but not verified (in case of need) (21L.A.43 (c));
- → <u>Before 1<sup>st</sup> flight</u> and approval of the flight conditions and issue of the permit to fly:
  - → EASA conducts 'a physical inspection and assessment of the aircraft' (21L.B.242 (a)(2));
  - $\rightarrow$  NAA conducts a 'physical inspection of the aircraft' (21L.B.241 (a)(2));

#### $\rightarrow$ The purpose is for:

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- → EASA to ensure that the aircraft is capable of safe flight, and that flight testing can be conducted safely;
- → The NAA to be satisfied that the aircraft conforms to the design for the requested permit to fly;







#### **Specifics of Part 21 Light -Declared**

- → The location is at the facilities of the declarant and the physical location of the aircraft that will be issued with the Permit to Fly;
- → Findings can be raised against the aircraft (21L.B.241 (c) & 21L.B.242 (b));
- → If there are no findings (or findings have been resolved) EASA approves the flight conditions and the NAA issues the Permit to Fly.







IFASA

#### **Specifics of Part 21 Light – Declared**

- → At the end of compliance demonstration all compliance documentation is provided to EASA (but not verified) (21L.A.43 (c));
- → The declarant submits a Declaration of Design Compliance stating that the design complies with the applicable Technical Specifications (21L.A.43 (a));
- $\rightarrow$  At this stage this declaration then triggers:
  - → EASA to conduct 'a physical inspection and assessment of the first article of that product in the final configuration' (21L.B.46 (c))
  - → The NAA to conduct 'a first article inspection' (21L.B.251 (b)& 21L.B.161 (c))





### Specifics of Part 21 Light –Declared

- → Purpose of 'first article inspection':
  - $\rightarrow$  For EASA to:
    - → Ensure that the designed aircraft is capable of conducting safe flight during in-service operations and does not have any environmental incompatibilities;
  - $\rightarrow$  For the NAA to:
    - → Conduct oversight of the production organization using Part 21 Light Subpart R;
    - → Ensure conformity of the aircraft for the first Restricted Certificate of Airworthiness (and Restricted Noise Certificate).

#### $\rightarrow$ Timing;

→ Should <u>not</u> be considered to be a single one day event and could be a series of visits before and after the declaration of design compliance (e.g. noise testing).



#### **Specifics of Part 21 Light - Declared**

- $\rightarrow$  Means of conducting the 'first article inspection':
  - → Gathering evidence to support compliance through a physical inspection and assessment of the aircraft and, if applicable, of the engine and the propeller;
  - $\rightarrow$  Other forms of evidence include:
    - → Witnessing or participating to live testing (including flight testing) of the aircraft, engine, propeller, systems or components;
    - $\rightarrow$  Review of the completeness of the compliance-demonstration plan;
    - → Determination of the completeness of supporting compliance documentation and test reports and how they relate to the first article under inspection;
    - $\rightarrow$  Discussions with key design and production personnel;
    - → In case of need and if relevant, a review of the design processes and procedures in order to determine root causes of any issues that are discovered

#### **Specifics of Part 21 Light -Declared**

- $\rightarrow$  If no findings are raised then:
  - $\rightarrow$  EASA registers the Declaration of Design Compliance (21L.B.63);
  - → The NAA issues the first Restricted Certificate of Airworthiness and Restricted Noise Certificate (21L.B.162 (b) & 21L.B.172 (a)).

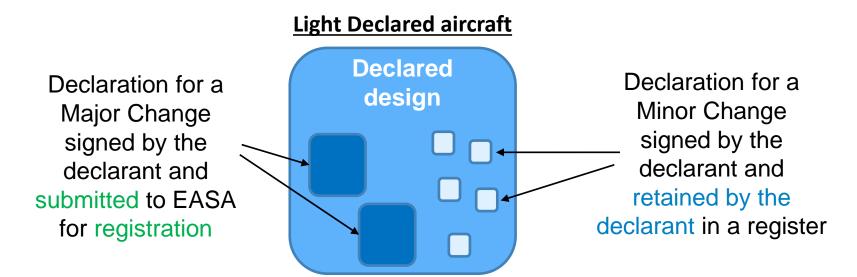




# Part 21 Light – Design Changes



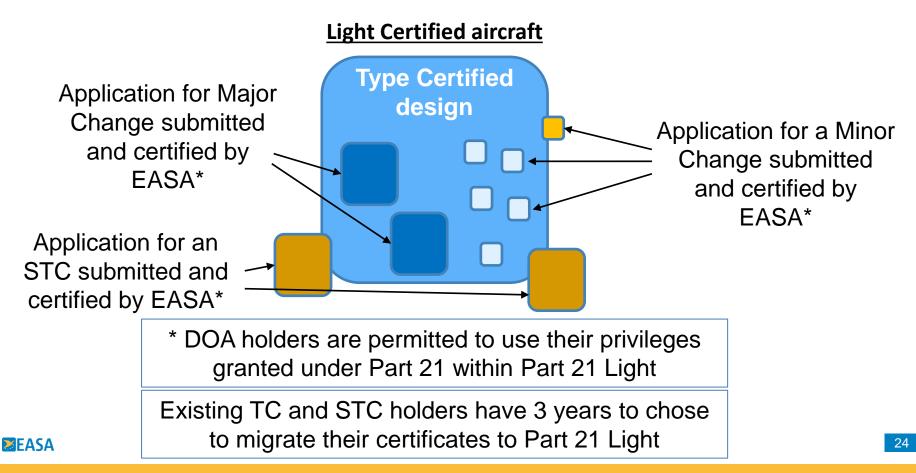
#### Part 21 Light - Design Changes and STCs (1 of 2)



Only the original declarant can make a declaration of design compliance for a major change (no STCs)

For minor changes, the original declarant or a DOA holder can make a declaration of design compliance

#### Part 21 Light - Design Changes and STCs (2 of 2)





# Part 21 Light –Conformity



### Part 21 Light - Conformity of individual aircraft

#### Statement of Conformity

EASA Form 52B issued by manufacturer to attest conformity of an aircraft.

#### Authorised Release Certificate

EASA Form 1 issued by manufacturer for engines, propellers and parts.

#### NAA Oversight

After first article inspection, further oversight visits to ensure confirmity are carried out after a risk based evaluation.



#### CofA/RCofA

Certificate of Airworthiness or Restricted Certificate of Airworthiness issued by NAA after ensuring conformity

#### Design data

Products and parts must be produced in conformity with approved or declared design data.

# Processes and procedures

Manufacturer must use established processes and procedures to ensure conformity to approved or declared design data.



# Part 21 Light – Safety issues and enforcement



#### Part 21 Light- Safety issues and enforcement

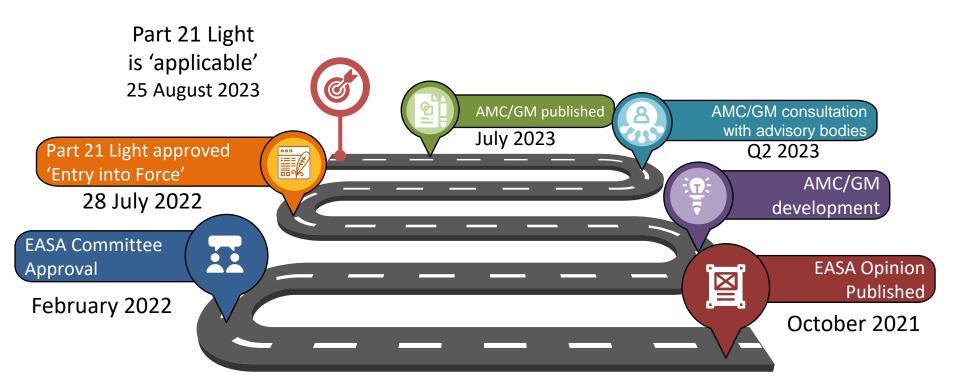
- → For safety issues, there is no change to the reporting obligations and the Airworthiness Directive process for certified or declared aircraft;
- → Findings can be raised against declared organisations and declared aircraft (especially before registration of declaration of design compliance);
- → Enforcement actions for certified products are unchanged but for declared aircraft this could include deregistration of the declaration of design compliance.



# Part 21 Light – Current Status



# Next steps for Part 21 Light



# **Part 21 Light AMC/GM status**

- $\rightarrow$  Direct consultation of AMC/GM 'packages' with the Advisory Bodies (GA COM/GA TeB) will be conducted (4 -6 weeks);
  - Package 1 Initial Airworthiness (Subparts A, B, C and P)  $\rightarrow$
  - Package 2 Design and Production Organisations (Subparts G, J and R)  $\rightarrow$
  - Package 3 Design Changes and Repairs (Subparts D, E, F, M and N)  $\rightarrow$
  - Package 4 Airworthiness and Noise Certificates and others (Subparts H, I, K and Q)  $\rightarrow$
- $\rightarrow$  AMC/GM will then be updated based upon comments;
- $\rightarrow$  Complete AMC/GM will be finalised and after proofreading will be circulated for management approval;
- $\rightarrow$  Aim to have the ED Decision before Part 21 Light applicability date.

## Part 21 Light – Webinar

Part 21 Light Webinar 15:00 to 17:00 on 27<sup>th</sup> April 2023 Registration Required SLIDO also available before Webinar





# Thank you for your attention!



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