

Abnormal Acceleration on Takeoff

EOFDM Conference 2016

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Safety Information Bulletin

Operations

SIB No.: 2016-02

Issued: 16 February 2016

Subject: Use of Erroneous Parameters at Take-off

Ref. Publications:

- U.S. National Air & Space Administration (NASA) Study [NASA/TM-2012-216007](#) "Performance Data Errors in Air Carrier Operations: Causes and Countermeasures".
- Australian Transport Safety Bureau (ATSB) Research and Analysis Report [AR-2009-052](#) "Take-off Performance Calculation and Entry Errors - A Global Perspective".
- Laboratory of Applied Anthropology, on behalf of BEA (Bureau d'Enquêtes et d'Analyses pour la sécurité de l'aviation civile) and DGAC France: [DOC AA 556/2008](#) "Use of erroneous parameters at take-off".
- EASA [AMC 20-25](#) "Airworthiness and Operational consideration for Electronic Flight Bags (EFB)".
- European Commercial Aviation Safety Team (ECAST) [European Operators Flight Data Monitoring](#):
 - Working Group A: "[Review of Accident Precursors for Runway Excursions](#)" and
 - Working Group B: "[Study for Runway Excursion Precursors](#)".
- European Authorities coordination group on Flight Data Monitoring ([EAFDM](#)) Document "[Developing standardised FDM-based indicators](#)".

Applicability:

Competent Authorities, Operators, Approved Training Organisations (ATO).

- Abnormal acceleration may not always be obvious to the crew
- Only the most serious cases get noticed and investigated
- Do we really understand the true scale of the problem?



Why not use FDM to get the full picture?

EOFDM Recommendation


| Increasing "proximity" to the accident  | | | |
|------------------------------------------------------------------------------------------------------------------------------|-------------------|---------------------------|-------------------------------------|
| Incorrect performance calculation | | No liftoff after rotation | |
| Inappropriate aircraft configuration | slow acceleration | Slow rotation | Reduced runway remaining at liftoff |
| GC out of limits | | Late rotation | |
| Reduced Elevator authority | | Engine power increase | |

Table 2: Precursor factors for accident scenario #2

WGA05 "Slow acceleration": Develop means to measure acceleration during the takeoff roll and detect abnormal values, taking in consideration the various factors that affect the takeoff performance.

Usefulness of “Abnormal Acceleration on Take-off” event:

For operators:

- Improved awareness of operational risk
- Opportunity to strengthen safety barriers (proactively)
- Data to support tactical/financial decisions about optional hardware

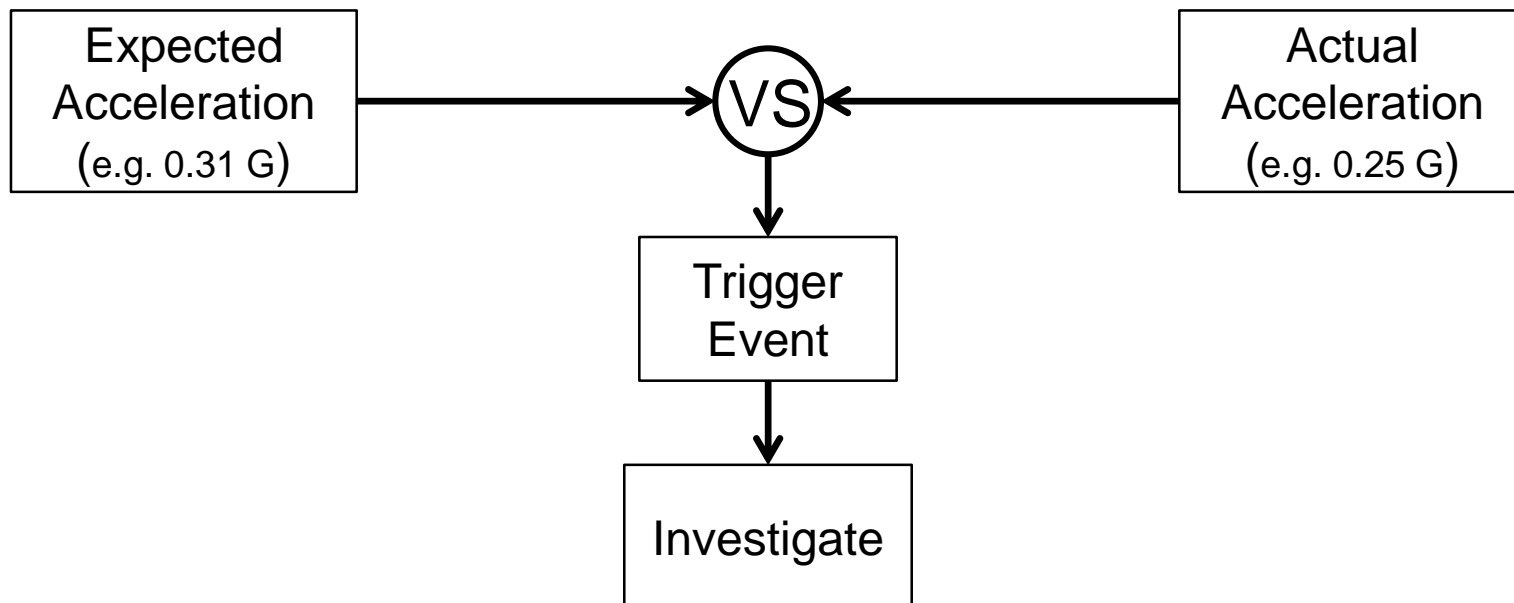
For regulators:

- Improved awareness of operational risk
- Promote adoption of best practices
- Data to support impact assessment of new regulation

Performance-related events are more challenging

- “Hard Landing” (trivial to implement)
- “Excessive speed on approach” (more complex)

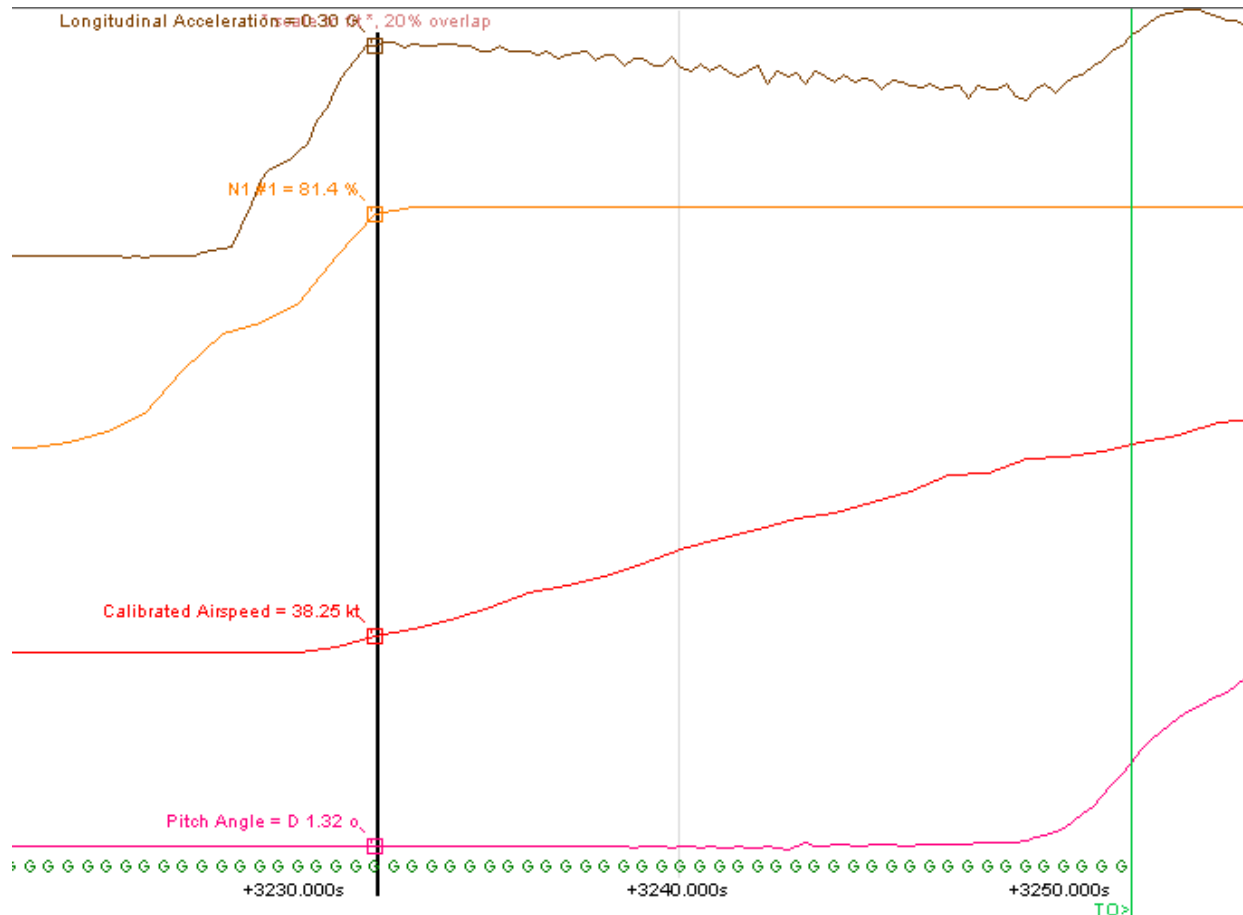
FDM event for ‘Abnormal Acceleration on Take-off’



- How can we predict the expected acceleration?
 - Acceleration is not constant during the T/O roll
 - Acceleration depends on many factors and is not documented in AFM / FCOM

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 - Acceleration is not constant during the T/O roll
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The typical acceleration profile



Peak Acceleration:
0.2 G ~ 0.4 G

Spool-up and
rotation techniques
are variable

Complex profile in
general, but simple
between “Power
set” and “rotation”

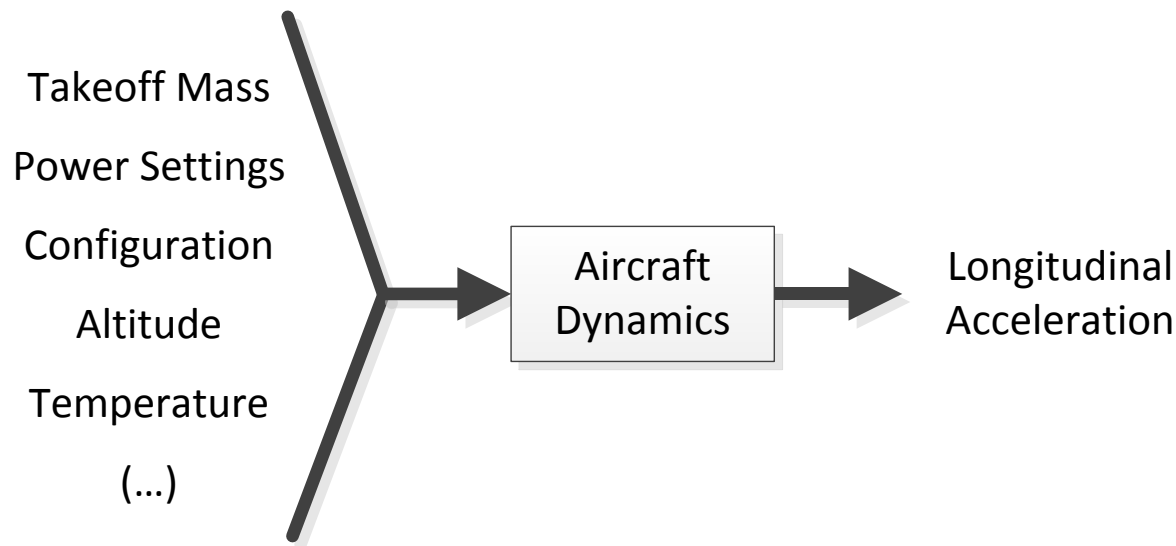
Use acceleration
@80kt:

- Rolling take-offs
- Far enough from
rotation

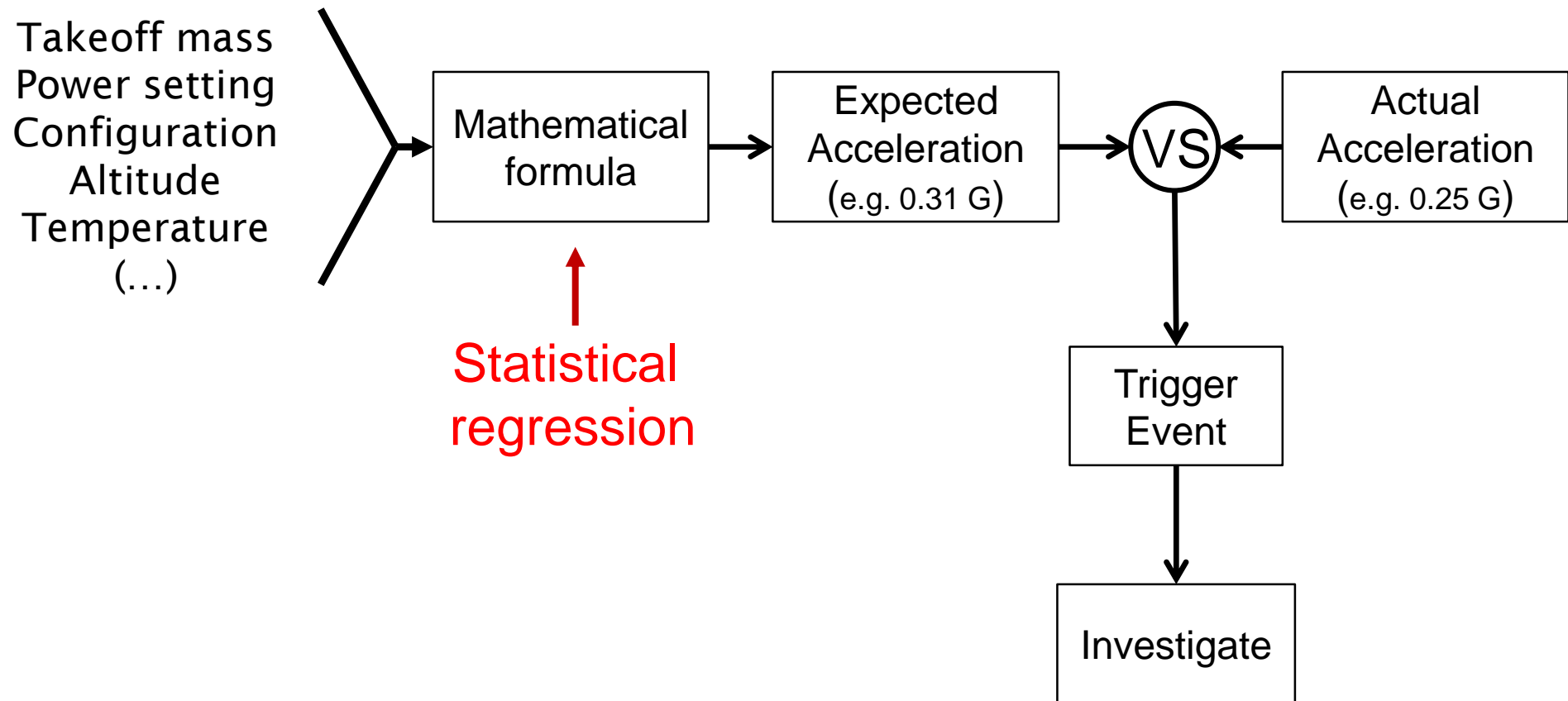
- How can we predict the expected acceleration?
 - Acceleration is not constant during the T/O roll
 - Acceleration depends on many factors and is not documented in AFM / FCOM

Between “Power Set” and “Rotation” @80kt:

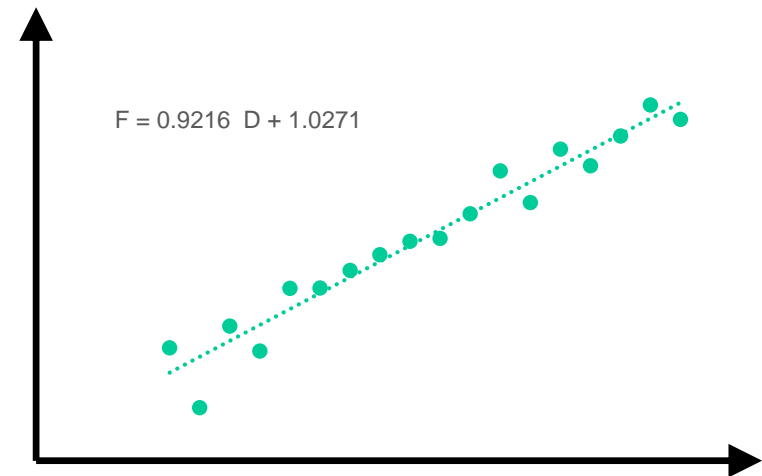
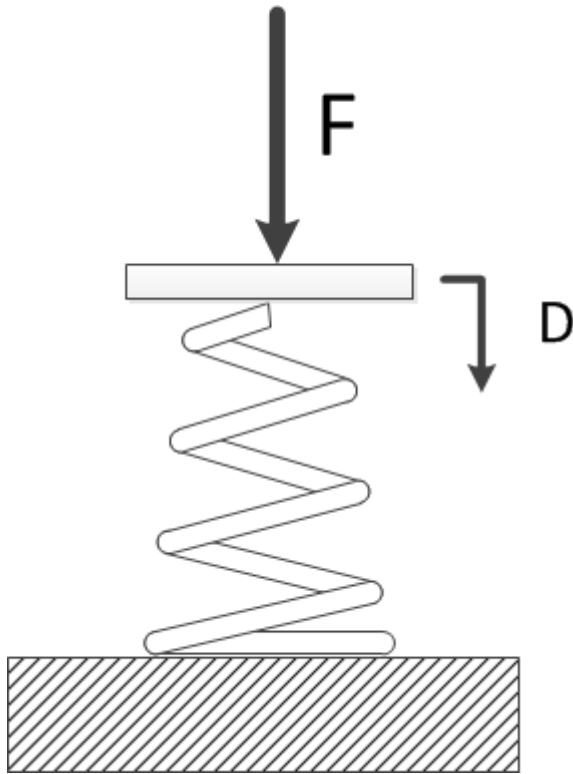
- acceleration is variable but solely dictated by physics (not pilot technique)
- We already have a basic idea about the most likely candidates



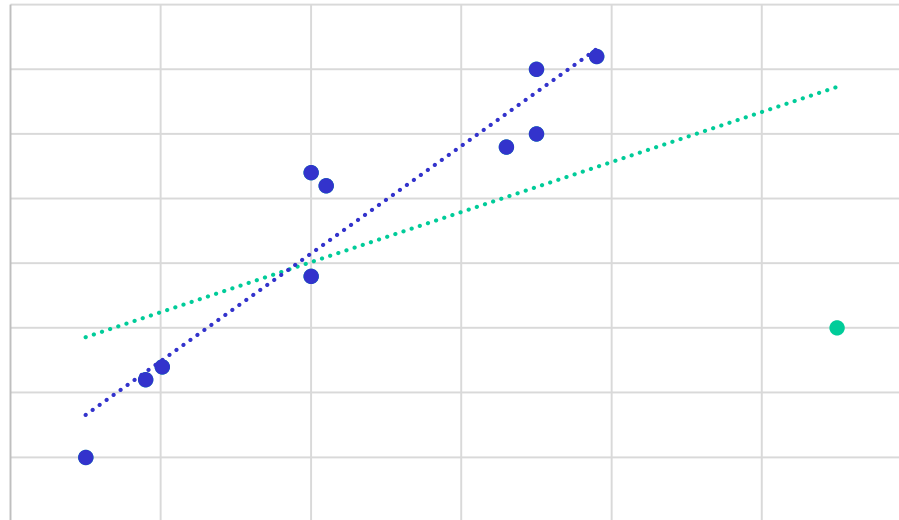
FDM event for 'Abnormal Acceleration on Takeoff'



Statistical regression is not “rocket science” ...



Statistical regression is not “rocket science” ... but it's not trivial either



- Bad data / outliers / sensor bias
- Selection of variables relevant for the model
- Assessment of model quality & refinements

Using statistical regression techniques it is possible to derive a mathematical model to calculate the expected acceleration

| WEIGHT | EPR | PALT | SAT | ACCEL | EXPECTED | DIFF |
|--------|------|------|------|--------|----------|--------|
| 29165 | 1.55 | -412 | 10.7 | 0.3432 | 0.3452 | 0.29% |
| 27998 | 1.56 | -32 | 18.1 | 0.3627 | 0.3655 | 0.78% |
| 36635 | 1.56 | -209 | 13.6 | 0.273 | 0.2684 | -1.72% |
| 33221 | 1.56 | 153 | 16.1 | 0.3015 | 0.298 | -1.15% |
| 28102 | 1.57 | 217 | 13.9 | 0.3677 | 0.3595 | -2.27% |
| 28484 | 1.56 | 165 | 22.1 | 0.3495 | 0.3575 | 2.24% |

The mathematical model only needs to be calculated once
(using historical flights)

... but does
it work in
practice?



Takeoff mass
Power setting
Configuration
Altitude
Temperature
(...)

Mathematical
formula

Expected
Acceleration
(e.g. 0.31 G)

VS

Actual
Acceleration
(e.g. 0.25 G)

Trigger
Event

Investigate

Methodology tested and successfully deployed for:

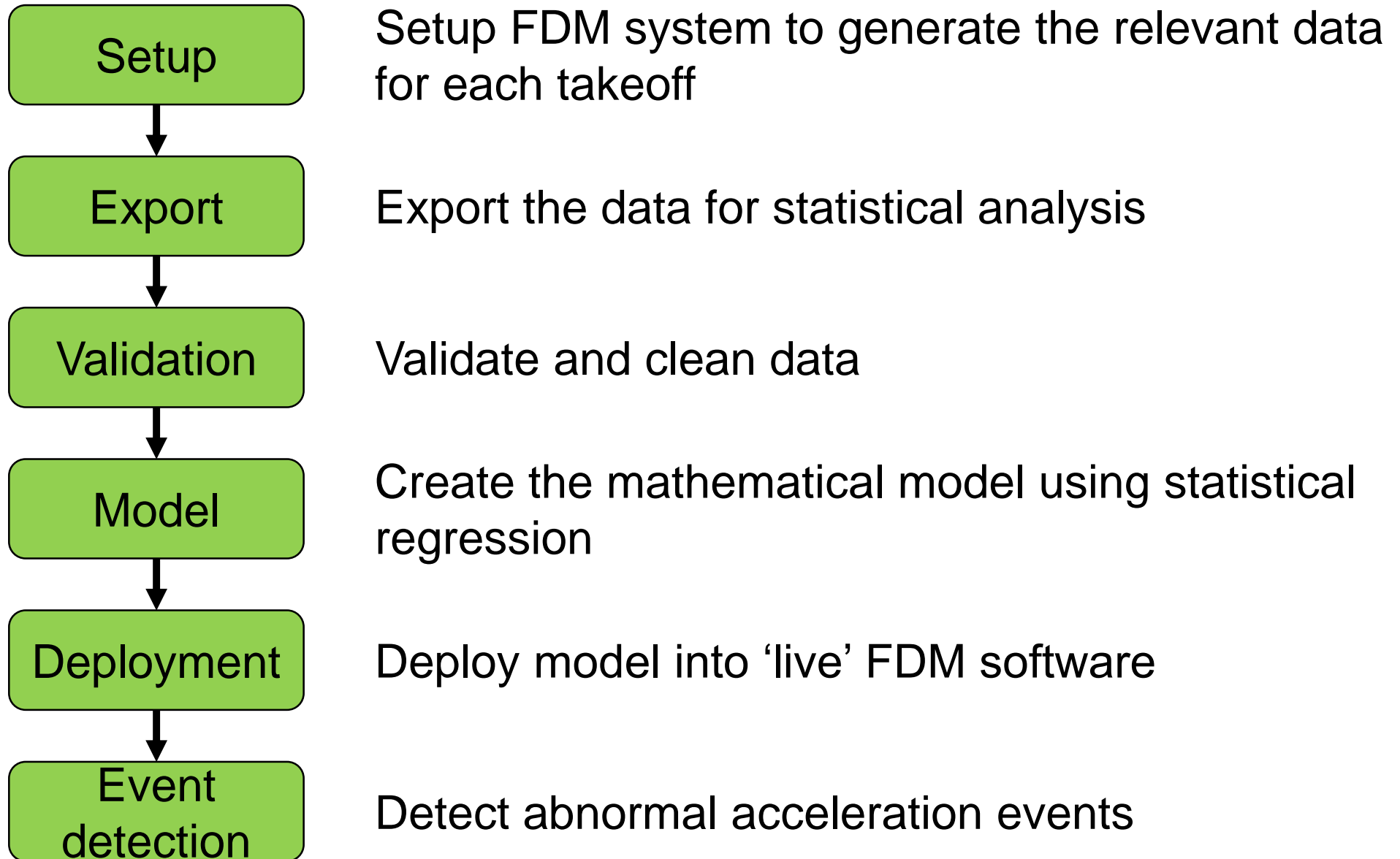
Gulfstream 550



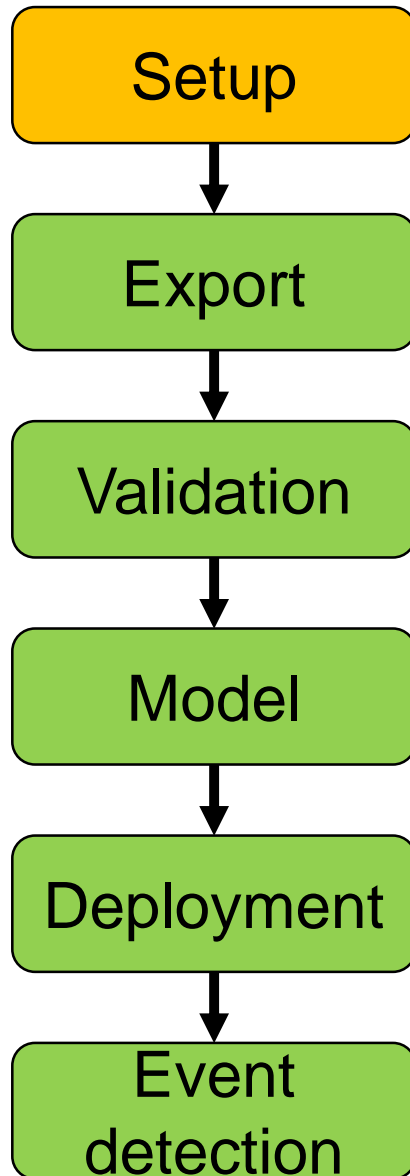
Airbus 320



Action plan



Action plan

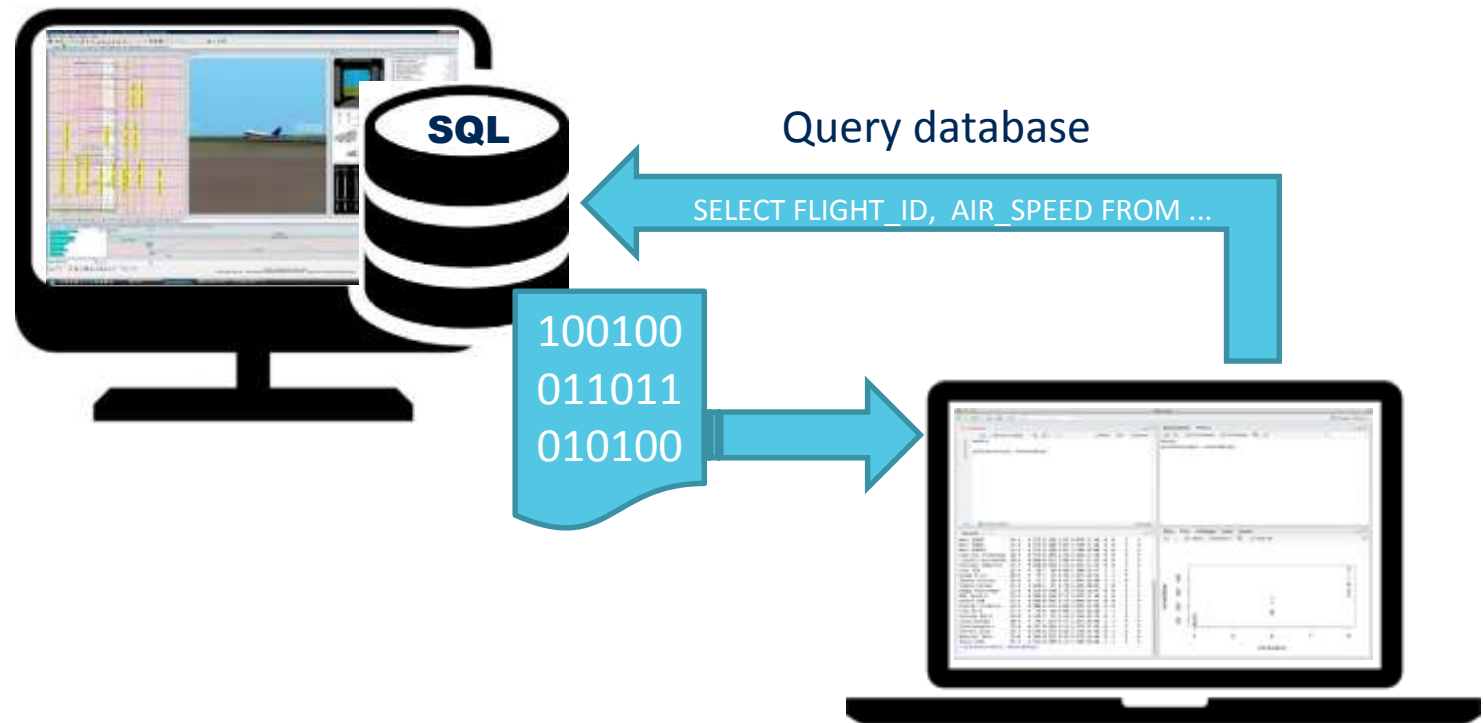
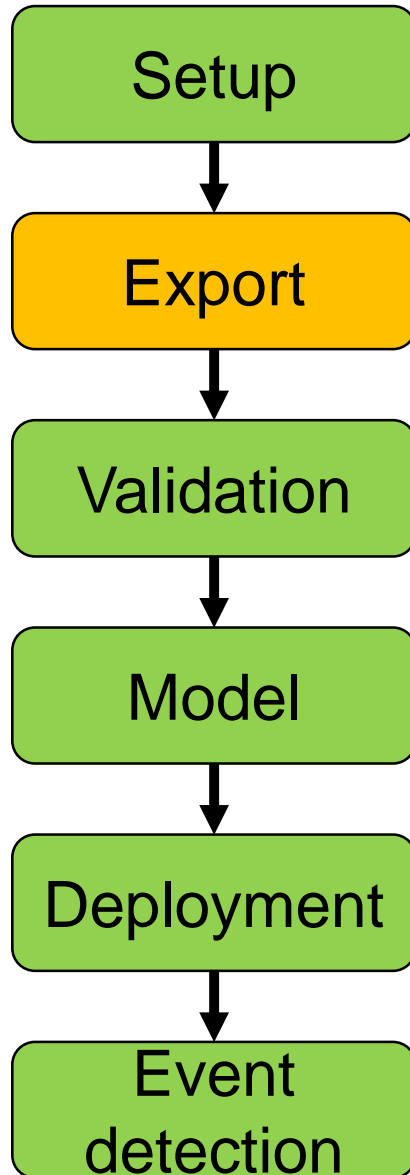


Setup FDM system to generate the relevant data for each takeoff

This may require configuration of the software as the model requires values on a specific instant in time

Action plan

Export the data for statistical analysis

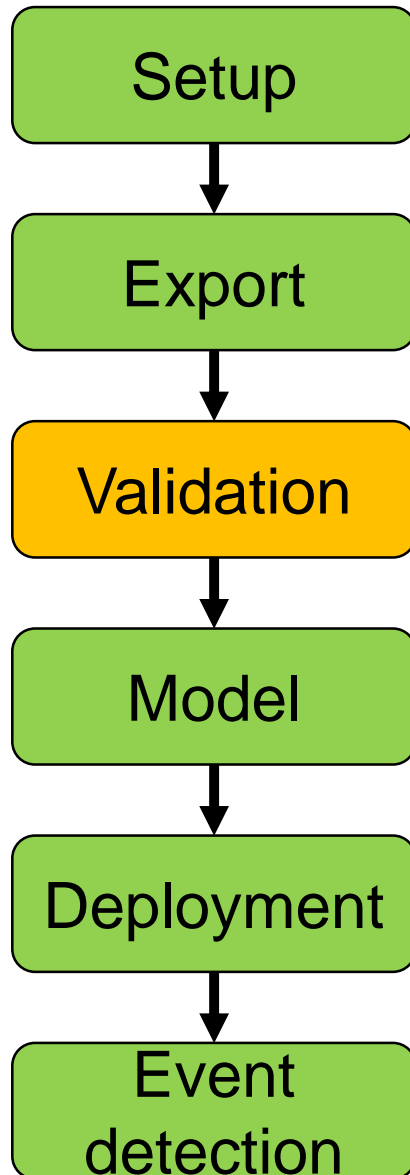


Action plan

Validate and clean the data

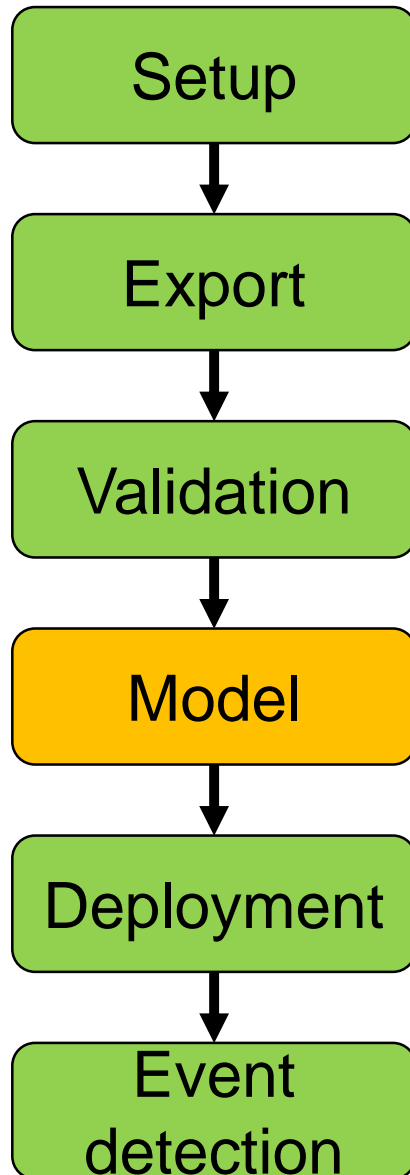
Find “weird” cases in input data

| ACCELAvg | TOW | PALT | TEMP | HEADW | N1 | N2 | EPR | FF | ICEENG | ICEWNG | BLEED | FLAP | STAB |
|----------|-------|------|------|-------|------|------|------|------|--------|--------|-------|------|------|
| 0.34 | 29231 | -412 | 10.7 | 9 | 90.7 | 93.8 | 1.54 | 2791 | 0 | 0 | 1 | 20 | -3.6 |
| 0.35 | 28072 | -319 | 13.5 | 11 | 88.4 | 93.4 | 1.50 | 2429 | 0 | 0 | 1 | 20 | -3.6 |
| 0.36 | 0 | 32 | 18.1 | 0 | 92.7 | 95.3 | 1.55 | 2861 | 0 | 0 | 1 | 20 | -3.6 |
| 0.29 | 27202 | 5791 | 16.8 | 7 | 84.1 | 92.1 | 1.40 | 2123 | 0 | 0 | 1 | 20 | -3.5 |
| 0.23 | 33830 | 393 | 18.6 | 6 | 84.7 | 92.4 | 1.41 | 2149 | 0 | 0 | 1 | 10 | -2.7 |



Action plan

Create the mathematical model using statistical regression



- Acceleration
- Take-off Mass
- Configuration
- Altitude
- Temperature
- Wind
- Bleed status
- Anti-ice status
- N1, N2, EPR
- Fuel Flow
- ...

| 0 | ACCELA | VG | TOW | PALT | TEMP | HEADW | N1 | N2 | EPR | FF | ICEENG | ICEWNG | BLEED | FLAP | STAB |
|---|--------|-------|------|------|------|-------|------|------|------|----|--------|--------|-------|------|------|
| | 0.34 | 29231 | -412 | 10.7 | 9 | 90.7 | 93.8 | 1.54 | 2791 | 0 | 0 | 0 | 1 | 20 | -3.6 |
| | 0.35 | 28072 | -319 | 13.5 | 11 | 88.4 | 93.4 | 1.50 | 2429 | 0 | 0 | 0 | 1 | 20 | -3.6 |
| | 0.36 | 28104 | -32 | 18.1 | 0 | 92.7 | 95.3 | 1.55 | 2861 | 0 | 0 | 0 | 1 | 20 | -3.6 |
| | 0.29 | 27202 | -147 | 16.8 | 7 | 84.1 | 92.1 | 1.40 | 2123 | 0 | 0 | 0 | 1 | 20 | -3.5 |
| | 0.23 | 33830 | 393 | 18.6 | 6 | 84.7 | 92.4 | 1.41 | 2149 | 0 | 0 | 0 | 1 | 10 | -2.7 |
| | 0.28 | 28936 | -121 | 17.3 | 6 | 84.2 | 92.2 | 1.40 | 2041 | 0 | 0 | 0 | 1 | 10 | -2.7 |
| | 0.27 | 36727 | -209 | 13.6 | 6 | 90.7 | 94.3 | 1.54 | 2598 | 0 | 0 | 0 | 1 | 20 | -3.6 |
| | 0.26 | 36733 | -139 | 26.1 | 7 | 91.0 | 95.7 | 1.51 | 2545 | 0 | 0 | 0 | 1 | 20 | -3.5 |
| | 0.30 | 26431 | 89 | 15.4 | 8 | 83.3 | 91.7 | 1.39 | 1988 | 0 | 0 | 0 | 1 | 10 | -2.7 |
| | 0.34 | 29039 | 386 | 16.6 | 19 | 90.6 | 94.5 | 1.53 | 2489 | 0 | 0 | 0 | 1 | 20 | -3.6 |
| | 0.36 | 27525 | 1183 | 17.4 | 8 | 93.1 | 95.4 | 1.57 | 2798 | 0 | 0 | 0 | 1 | 10 | -2.7 |



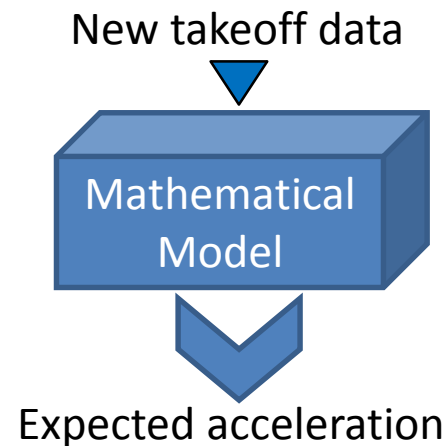
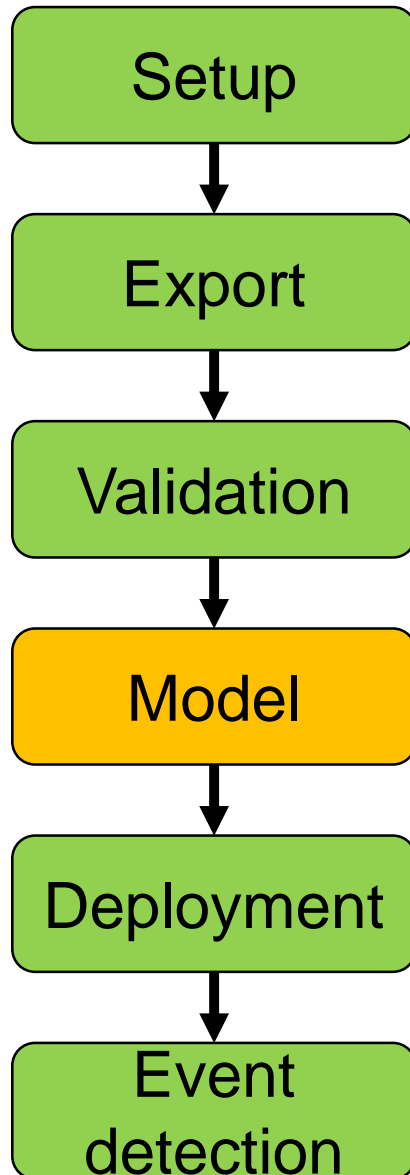
$$\begin{aligned}
 &Accel_{pr} \\
 &= \left(\frac{3549}{WEIGHT} \right) - (1.3 \times 10^{-5} \times PALT) - (1.5 \times 10^{-3} \times SAT) \\
 &\quad + (9.7 \times 10^{-3} \times EPR) + 0.9417
 \end{aligned}$$



Not all inputs are useful or relevant for the model

Action plan

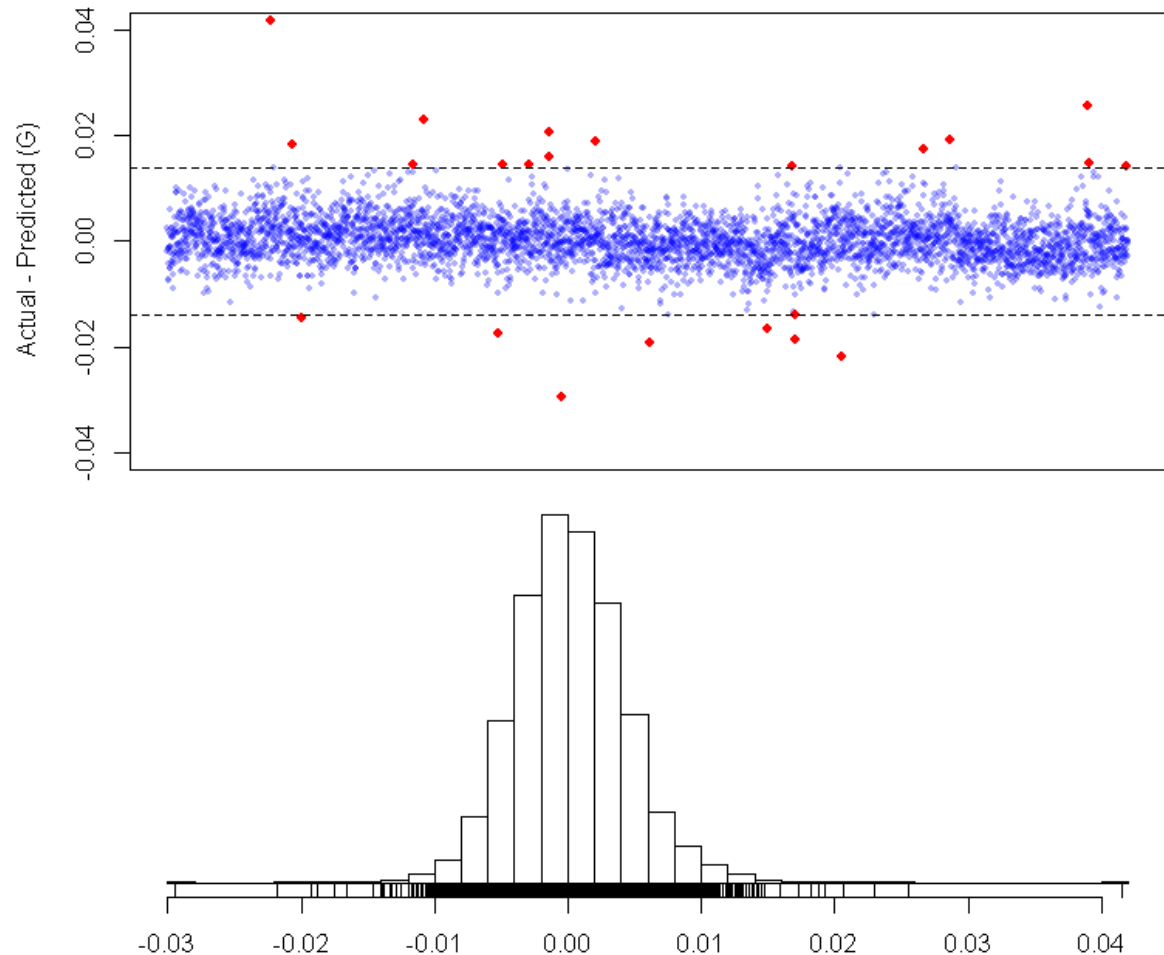
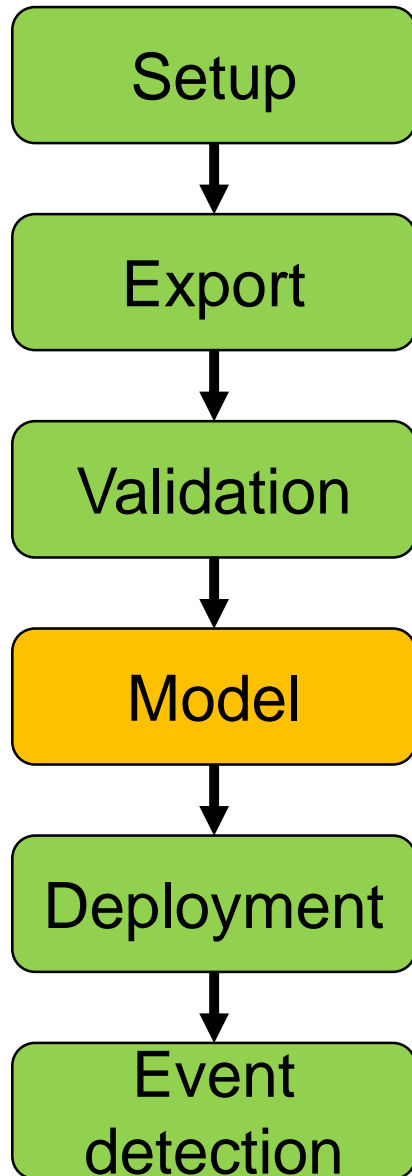
Create the mathematical model using statistical regression



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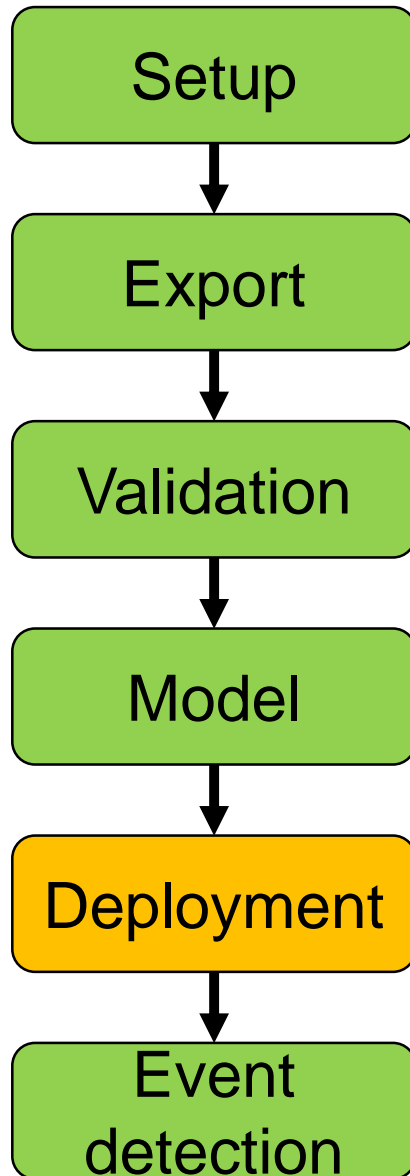
Action plan

Create the mathematical model using statistical regression



Action plan

Deploy model into 'live' FDM software



Edit State Definition - 'Takeoff_Model_01'

Name: Takeoff_Model_01 Effective from: 01 Jan 00 Status: Live

Description: estimate expected acceleration at 80kt on takeoff Last Modified: 02 Mar 16 15:26:57

Do NOT make changes unless you REALLY understand what you are doing!

Copy Verify ☒ Use context sensitive scripts?

Range Start Intermediate End Definitions Output Relevance Dependencies Test

Intermediate Script (executed after start of state, until end has been detected)

```
binPALT.CLEARTOMARKER();

binPWR.SETMARKERAT(var80KtInstant-1);
binPWR.CLEARTOMARKER();

binSAT.PRINT();
binSAT.SETMARKERAT(var80KtInstant-1);
binSAT.CLEARTOMARKER();
binSAT.PRINT();

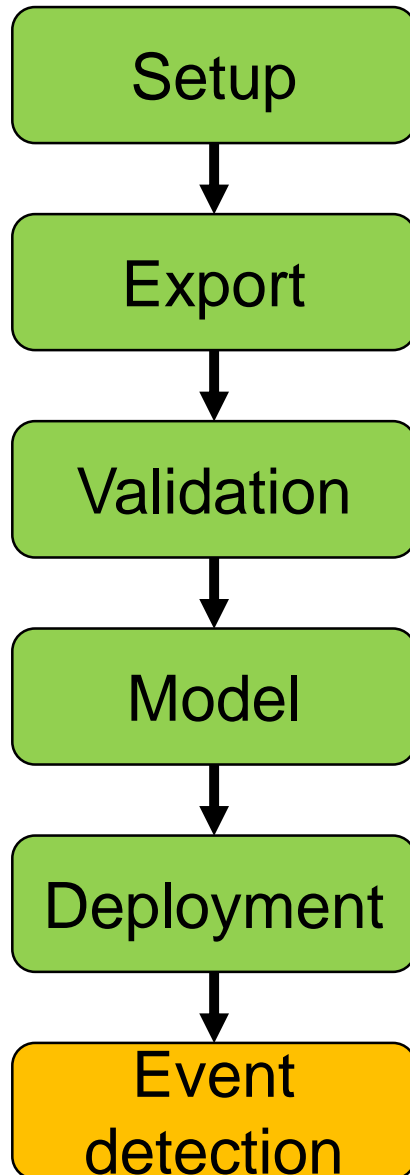
varPitch=binPITCH.GETAVERAGE();
varWeight=binWEIGHT.GETAVERAGE();
varFlap=binFLAP.GETAVERAGE();
varPitch=binPITCH.GETAVERAGE();
varLonG=binLonG.GETAVERAGE();
varPALT=binPALT.GETAVERAGE();
varPWR=binPWR.GETAVERAGE();
varSAT=binSAT.GETAVERAGE();

IF(varWeight <> 0){
  IF(varPALT > 3000){
    varLonG2=(-0.7313037)+(9736.773/varWeight)+(-0.00001242622*varPALT);
    varLonG2= varLonG2 +(0.00009512565*varSAT)+ (0.4803982*varPWR) + (0.003742431*varPitc
  }ELSE{
    IF(varPWR<1.47){
      varLonG2=(-0.978559)+(8646.972/varWeight)+(-0.00001212504*varPALT);
      varLonG2= varLonG2 +(0.0001641910*varSAT)+ (0.6752905*varPWR) + (0.002797335*varF
    } ELSE {
      varLonG2=(-0.8513354)+(11319.15/varWeight)+(-0.00001421565*varPALT);
      varLonG2= varLonG2 +(0.0001467673*varSAT)+ (0.5279223*varPWR) + (0.002271791*varF
    }
  }
  varResidual=varLonG-varLonG2;
}ELSE{
  varResidual=-1;
  varLonG2=0;
}

varResidual.PRINT();
binResidual.ADD(varResidual);
varPitch.PRINT();
```

Action plan

Detect abnormal acceleration events



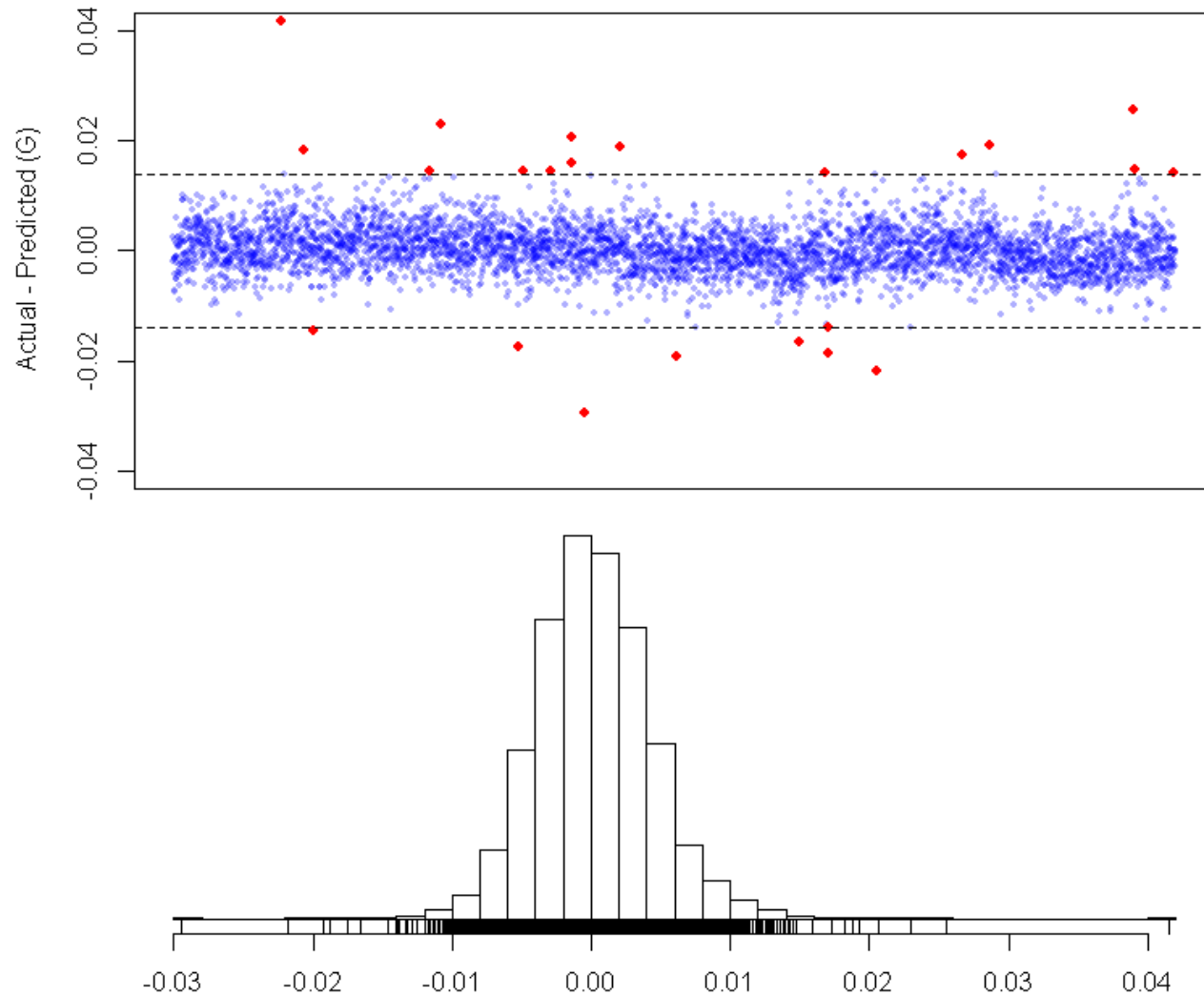
| | |
|--------------------------|-------|
| Total Fuel (KG) - FMS1 | 17367 |
| Total Fuel (KG) - FMS2 | 17367 |
| Total Fuel (KG) - FMS3 | 17367 |
| Low Takeoff Acceleration | 27% |
| LongG Residual @80kt | -0.10 |

| | |
|---------------------------|-------|
| Total Fuel (KG) - FMS1 | 10414 |
| Total Fuel (KG) - FMS2 | 10414 |
| Total Fuel (KG) - FMS3 | 10414 |
| High Takeoff Acceleration | 25% |
| LongG Residual @80kt | 0.03 |



Investigate

Findings



Thank you

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NETJETS®

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