

# Considerations for the Expanding Use of Computational Materials Capabilities in Additive Manufacturing

***Presented at:***

2021 EASA – FAA Workshop on Q&C of AM  
***Virtual***

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# Outline

- **ICME / CM – background and motivation**
- **Regulatory landscape**
- **Industry trends**
- **Metal AM as a use case**
- **V&V framework as a key enabler**
- **Overview of CM<sup>4</sup>QC Steering Group**
- **Summary**

Note: *in the context of this presentation*, the terms **CM** (Computational Materials) and **ICME** (Integrated Computational Materials Engineering) are used *interchangeably*

# ICME as ~~Emerging~~ Technology *Evolving*

## Commonly identified **benefits**:

- Cost savings
- Novel fit-for-purposes materials
- Integrated design, certification, and flexible manufacturing
- Risk reduction (*program risk* vs. *product safety risk*)

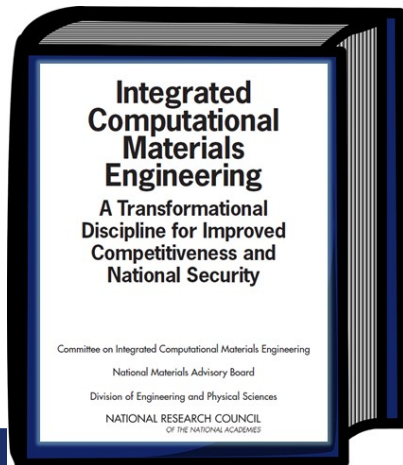
### AIM Program Overview

**DARPA AIM**



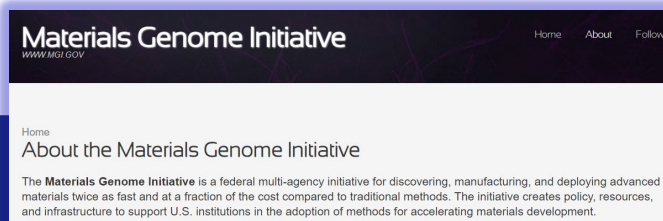
The AIM program initiative created a new materials development methodology that accelerates the insertion of new materials in order to achieve parity with the engine/platform development/design cycles. Accomplishments of the AIM program include:

- Establish design-driven material requirements by tightly coupling design and materials activities and tools.
- Providing earlier information (with confidence bounds) to designers throughout the development cycle.
- Controlling the performance, producibility, and cost of materials.
- Reducing risks of new material insertion risk while also decreasing costly, time-consuming data generation.
- Creating a knowledge base and tool kit for designers that links with computational design tools.



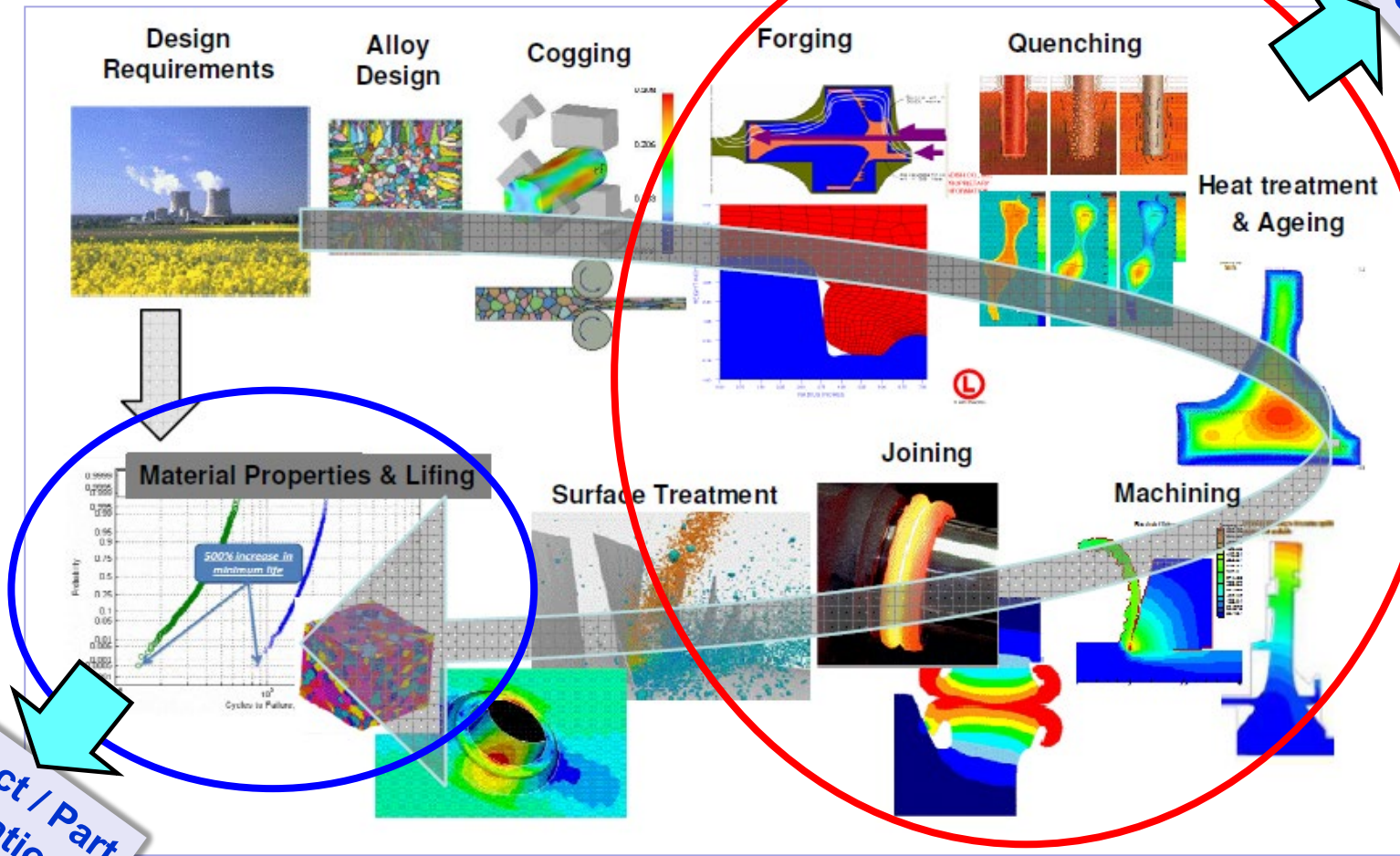
*Keeping pace with  
industry needs*

*Requires significant  
maturation to realize  
this benefit*



# Example: ICME Framework for Forged Components

Material & Process Specs



**Reference:** M. Glavicic et al., "Application of ICME to Turbine Engine Component Design Optimization", AIAA 2011-1738

# 14 CFR Part 25 Regulations - Materials

## *(Transport Category Aircraft)*

### **§ 25.603 Materials**

- The suitability and durability of materials used for parts, the failure of which could adversely affect safety, must —
  - a) Be established on the basis of experience or tests;***
  - b) Conform to approved specifications (such as industry or military specifications, or Technical Standard Orders) that ensure their having the strength and other properties assumed in the design data; and
  - c) Take into account the effects of environmental conditions, such as temperature and humidity, expected in service.

### **§ 25.605 Fabrication Methods**

- a) The methods of fabrication used must produce a consistently sound structure. If a fabrication process (such as gluing, spot welding, or heat treating) requires close control to reach this objective, the process must be performed under an approved process specification.
- b) Each new aircraft fabrication method ***must be substantiated by a test program.***



# 14 CFR Part 25 Regulations - Materials

(Transport Category Aircraft)

## § 25.613 Material Strength Properties and Design Values

- a) Material strength properties must be **based on enough tests** of material meeting approved specifications **to establish design values on a statistical basis**.
- b) Design values must be chosen to minimize the probability of structural failures due to material variability.
- d) The strength, detail design, and fabrication of the structure **must minimize the probability of disastrous fatigue failure**, particularly at points of stress concentration.
- e) Greater design values may be used if a “premium selection” of the material is made in which a **specimen of each individual item** is **tested before use**.

***No Allowance for Modeling or Analysis***

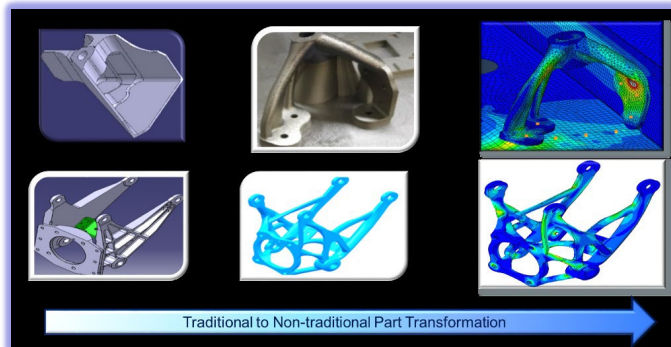
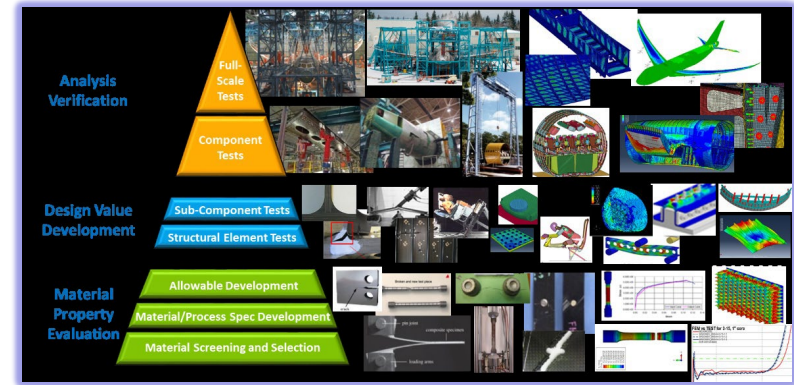
# Examples of “Model-Friendly” Domains

- Proof of Structure → Part 25 (14 CFR 25.307)
  - Structural analysis may be used *only if the structure conforms to that for which experience has shown this method to be reliable*
- Damage Tolerance → Part 25 (AC 25.571-1D)
  - In general, “analysis supported by test evidence” is accepted
- Damage Tolerance → Part 33 (AC 33.70-1)
  - Analysis is accepted (e.g. stress, heat transfer, crack growth, ... )
    - However, “...*the analysis approach should be validated against relevant test data*”
- AC 20-146 “Methodology for Dynamic Seat Certification by Analysis” → Parts 23, 25, 27, 29
  - Needs to be validated by test
  - One of the few examples of “certification by analysis” (CbA)
  - *Rational Analysis* - an analysis based on good engineering principles, judgment, and/or *accepted methodology* (AC 25.562-1b)

# Industry Trends

## Example - “Smarter Testing”

**“Use of advanced analysis techniques using fundamental (coupon-derived) inputs can lead to reduced quantities of programmed mid-level structural tests, reducing airplane development costs and risks”.**



**“...AM presents new challenges for certification in that there are no traditional validated analysis methods suited to the arbitrary and organic nature of many AM parts...”**

**Reference:** S. Chisholm et al, “Smarter Testing Through Simulation for Efficient Design and Attainment of Regulatory Compliance”, Boeing, Presented at 30<sup>th</sup> ICAF Symposium – Kraków, 5 – 7 June 2019.



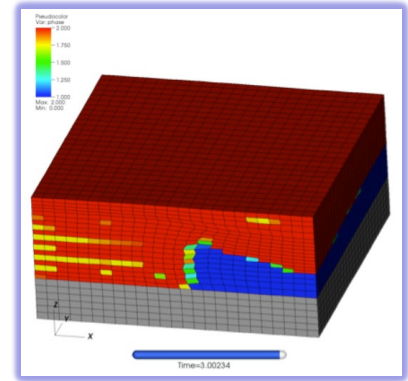
# ***AM as a Use Case*** for Developing **Model-based Qualification Framework**

- **Relevance to other material systems' attributes**  
(*casting, welding, powder metallurgy, ...*)
- **Highly complex “eco system”** (process → microstructure → properties)
- **Pathway to future technologies (e.g. UAS/UAM) and applications:**
  - Topologically optimized structures
  - Location specific / gradient microstructures
  - Multi-material systems
  - Multi-functional systems (e.g. embedded electronics)

# Modeling as an Enabler for Q&C of AM

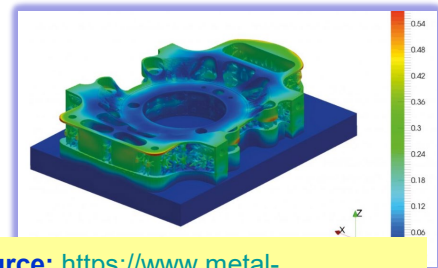
Physics-based process models have been identified as being foundational to qualification of additively manufactured metal parts.

Ref: W. King, “Accelerated Certification for Additively Manufactured Metals”, LLNL, 2015.



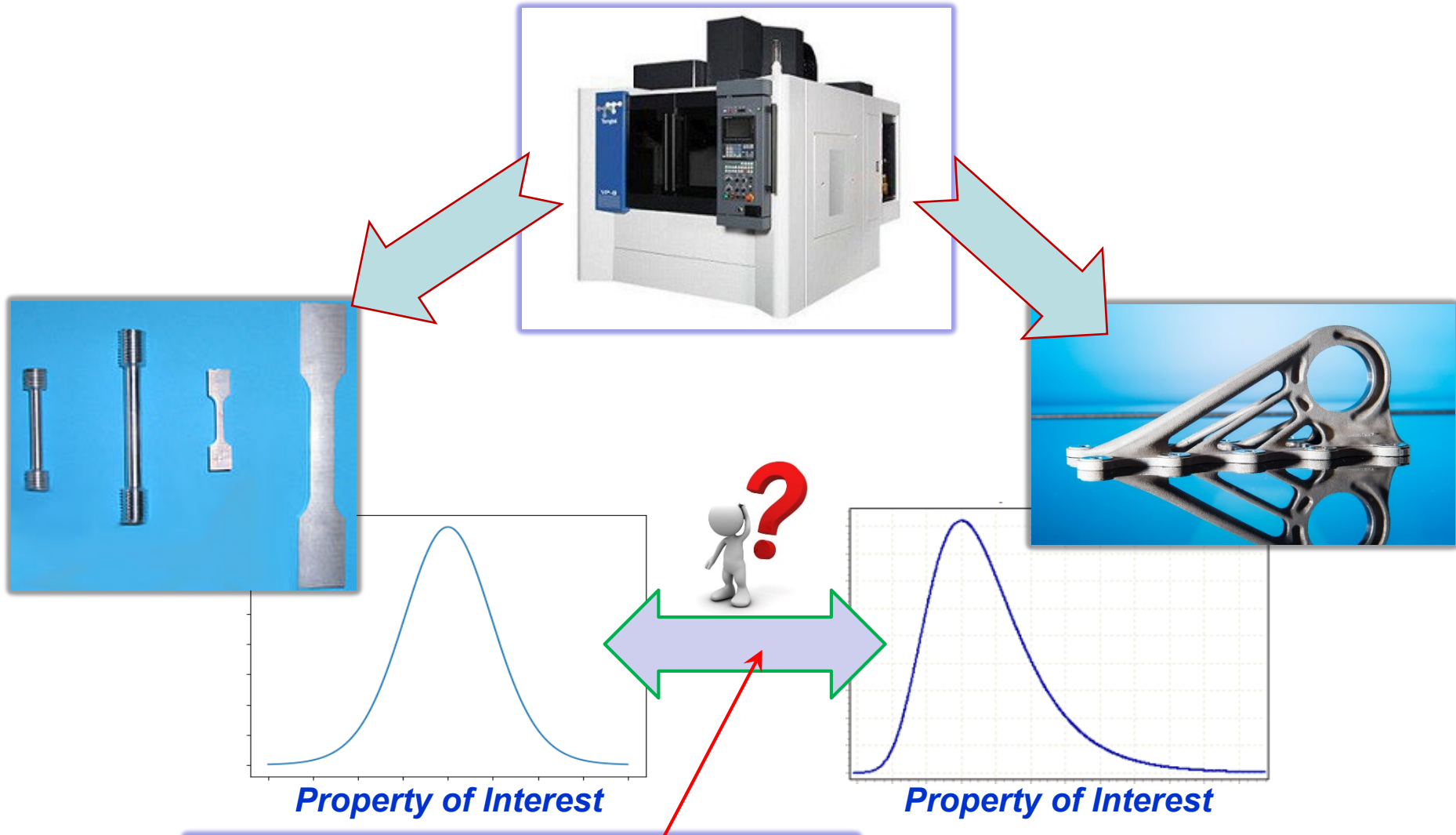
## *Examples of CM application areas for AM:*

- Process parameters optimization
- In-situ monitoring algorithms
- Prediction of distortion
- Effect of defects on part's durability
- Correlation between coupon-level and part-level properties → *see next slide*



Source: <https://www.metal-am.com/articles/distortion-in-metal-3d-printing-modelling-and-mitigation/>

# Example: Part vs. Coupon Properties



This understanding can be enabled  
by physics-based CM models

# Multi-Scale Framework Considerations

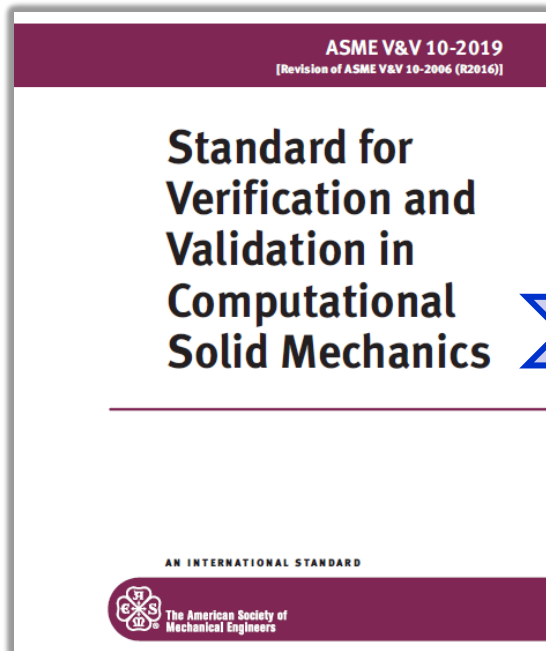


- Not everything has to be (*or can be...*) derived from the first principles
  - A meaningful combination of physics-based and *empirical* models can be used as a maturation path
  - “Big Data” / ML may provide a complementary approach
- Models validation is key → **V&V and UQ**
  - But the level of effort can sometimes overshadow the conventional characterization approach...
- “Meso” attributes can be used to streamline process control. *Examples:*
  - Use of microstructure attributes to control part’s properties
  - Controlling melt pool in AM process

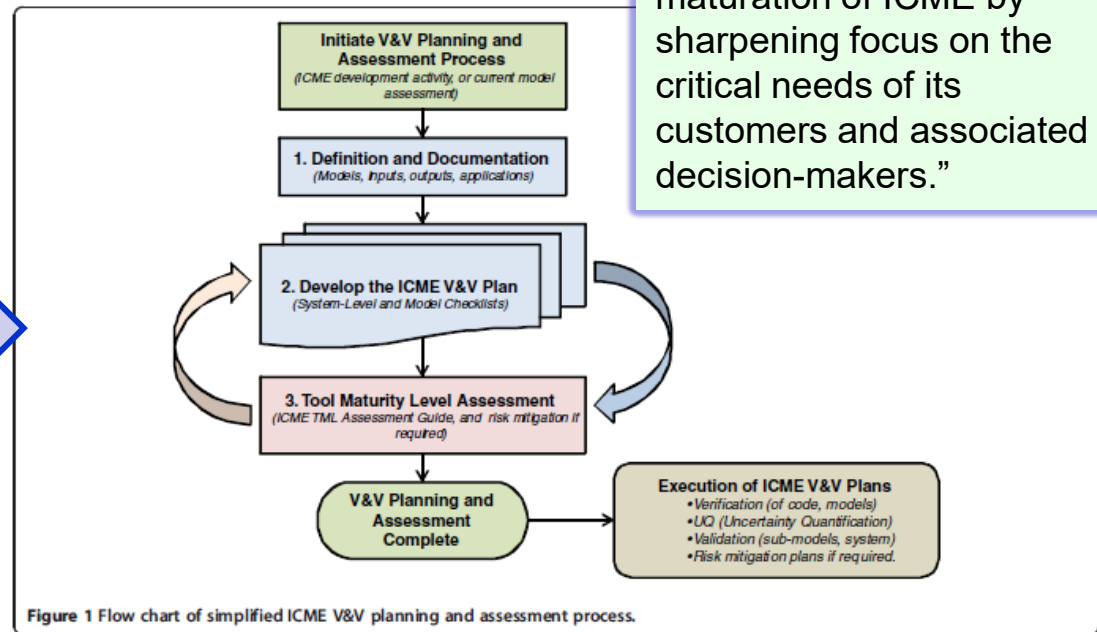
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# V&V Framework for ICME

## General V&V Framework (for Computational Solid Mechanics)



## V&V Framework Tailored to ICME



Ref: B. Cowles et al, “*Verification and validation of ICME methods and models for aerospace applications*”, Integrating Materials and Manufacturing Innovation 2012 (<http://www.immijournal.com/content/1/1/2> )

sponsored by AFRL / USAF



# NASA / NIST / FAA Technical Interchange Meeting (**TIM**) on Computational Materials Approaches for Qualification by Analysis for Aerospace Applications

- Held at NASA Langley Research Center on January 15-16, 2020.
- Motivated by three related factors:
  - The **aerospace industry's increasing interest** in expanding the use of computational materials for Q&C of process-intensive metallic materials.
  - The **rapid maturation of computational materials capabilities** across a range of applications.
  - A **general lack of coordination** of development and investment in these capabilities by funding organizations.
- Included 60 subject matter experts (SMEs) representing 8 aerospace manufacturers, 7 government organizations and 2 universities.
- Key **objectives** were to:
  - Understand existing gaps in model-based, e.g., computational materials, capabilities for processing and performance prediction for aerospace materials and components.
  - Forecast how capabilities can be matured to support material, process and part-level Q&C.

# Development of Computational Materials (CM) Capabilities for Metal AM

**Co-organizers: NASA and FAA**

NASA/TM-20210015175

DOT/FAA/TC-20/38



NASA / NIST / FAA Technical Interchange Meeting on Computational Materials Approaches for Qualification by Analysis for Aerospace Applications

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**Membership**



## Government

NIST  
AFRL  
Sandia NL  
NAVAIR  
ORNL  
Army Aviation  
NASA  
FAA

## Industry

Boeing  
Lockheed-Martin / Sikorsky  
Raytheon / P&W  
GE Aviation  
Spirit Aerosystems  
Honeywell Aerospace  
Howmet Aerospace  
SwRI  
Northrup-Grumman  
Textron Aviation / Bell

## Academia

Carnegie Mellon  
UTSA  
Vanderbilt  
Penn State  
Northwestern

**CM4QC SG formed per recommendations of the Jan. 2020 TIM**

# Goals of the CM4QC Steering Group

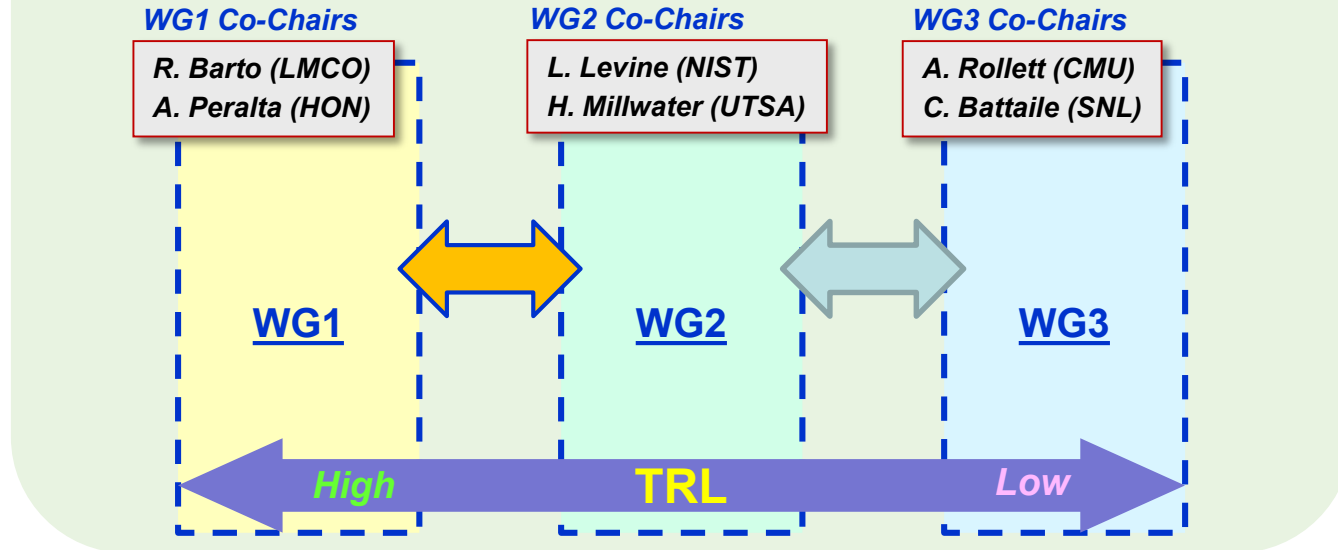
- To **inform** U.S. industry and the U.S. government **regarding the R&D investment opportunities** toward development of CM-based approaches for qualification and certification (Q&C) of process intensive metallic materials (PIM)
  - Initial focus is on powder bed fusion (PBF)
  - Subsequent consideration of wire directed energy deposition (DED) and powder DED.
- To identify key **considerations and enablers required to increase airworthiness / certifying authorities' acceptance** of computational methods use for Q&C of structural or flight-critical PIM parts
- To **increase dialogue among the stakeholder organizations**, develop a common understanding of the state-of-the-art of CM in the Q&C domain including related gaps and challenges.
- To seek **opportunities for sharing capabilities, methods, tools, codes, best practices** and discussion of regulatory considerations.

# CM4QC Org Chart

## SG Co-Chairs

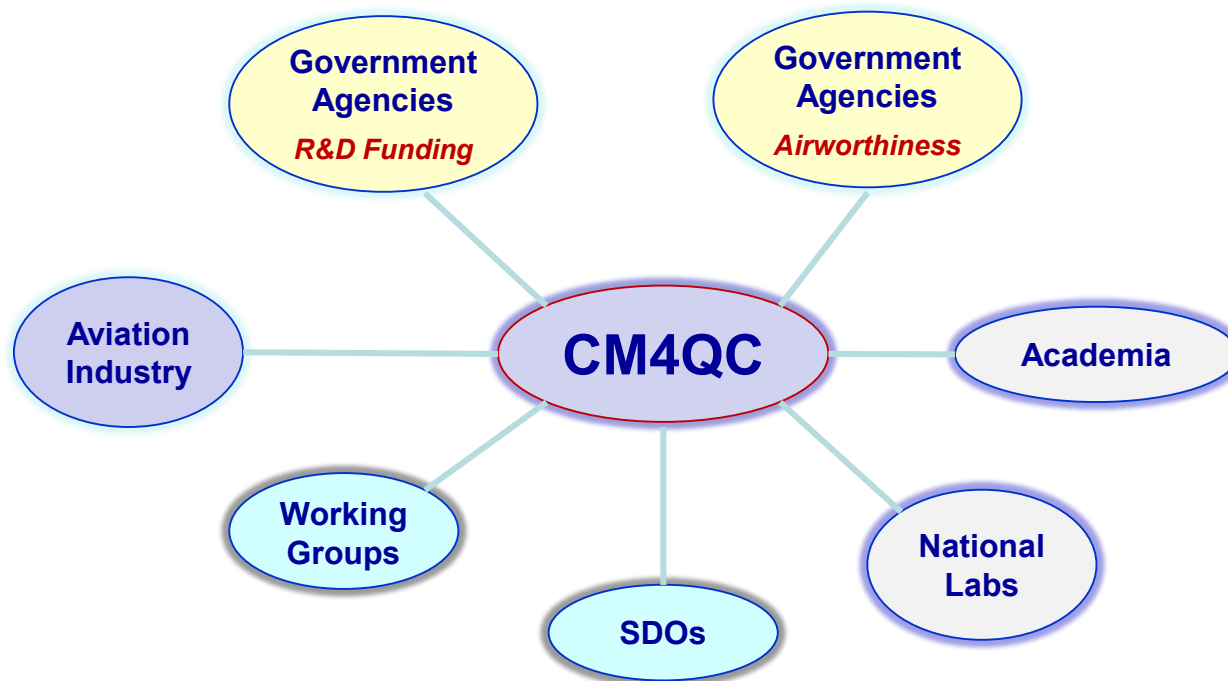
E. Glaessgen (NASA)  
M. Gorelik (FAA)

## CM4QC Steering Group



- **WG 1:** Understanding industry priorities / timeline and key regulatory considerations
- **WG 2:** Strategies for maturation and transition of Research to Engineering
- **WG 3:** Development of required computational materials and measurement capabilities

# Key Stakeholders for CM4QC SG





# Development of CM Roadmap

- Key output of the current phase of CM<sup>4</sup>QC activities
- Target completion date – mid-2022 (*estimate*)
- **Examples of the Roadmap topics:**    *- preliminary -*
  - *Industry's vision for CM adoption*
  - *Identification of key CM and enabling technologies*
  - *Key elements and associated methods for CM V&V framework*
  - *Technology maturation path*
  - *“State of industry” assessment of CM tools*
  - *Considerations for acceptable levels of V&V (regulatory perspective)*
  - *Key elements of the CM Eco System's*



# Summary

- Gradual maturation of ICME / CM is a good path forward
  - Strong interest from industry, supported by a technical and business case
  - Demonstrated **early successes** *outside of regulatory domain* (e.g. material & process development and optimization, preliminary design)
  - Longer-term – increasing use in Q&C domain
- Key requirements for maturation of CM: **UQ and V&V**
  - A heavily data-driven process
- Metal AM as a “use case” for CM
- Complex multi-disciplinary problem → ***importance of inter-agency and industry-government-academia collaboration, and engagement with SDOs***

# Discussion



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