Issue: 03



Date: 18 August 2021

TYPE-CERTIFICATE DATA SHEET

No. P.033

for Propeller MTV-37 Series

Type Certificate Holder MT-Propeller Entwicklung GmbH

> Flugplatzstraße 1 94348 Atting Germany

For Models: MTV-37-1



Date: 18 August 2021

TCDS No.: P.033 Issue: 03

Intentionally left blank



Date: 18 August 2021

TCDS No.: P.033 Issue: 03

TABLE OF CONTENTS

I. General	. 4
1. Type / Models	. 4
2. Type Certificate Holder	. 4
3. Manufacturer	. 4
4. Date of Application	. 4
5. EASA Type Certification Date	. 4
II. Certification Basis	
1. Reference Date for determining the applicable airworthiness requirements	. 4
2. EASA Certification Basis	. 5
2.1. Airworthiness Standards	. 5
2.2. Special Conditions (SC)	. 5
2.3. Equivalent Safety Findings (ESF)	. 5
2.4. Deviations	. 5
III. Technical Characteristics	. 5
1. Type Design Definition	. 5
2. Description	
3. Equipment	. 6
4. Dimensions	. 6
5. Weight	
6. Hub / Blade Combinations	
7. Control System	. 6
8. Adaptation to Engine	. 6
9. Direction of Rotation	. 6
IV. Operating Limitations	. 7
1. Approved Installations	
2. Maximum Take Off Power and Speed	
3. Maximum Continuous Power and Speed	. 7
4. Propeller Pitch Angle	. 7
V. Operating and Service Instructions	. 7
VI. Notes	. 8
SECTION: ADMINISTRATIVE	10
I. Acronyms and Abbreviations	
II. Type Certificate Holder Record	10
III. Change Record	10

Date: 18 August 2021

TCDS No.: P.033 Issue: 03

I. General

1. Type / Models

MTV-37 / MTV-37-1

2. Type Certificate Holder

MT-Propeller Entwicklung GmbH Flugplatzstraße 1 94348 Atting Germany

Design Organisation Approval No.: EASA.21J.020

3. Manufacturer

MT-Propeller Entwicklung GmbH

4. Date of Application

MTV-37-1: 14 October 2016

5. EASA Type Certification Date

MTV-37-1: 09 December 2019

II. Certification Basis

1. Reference Date for determining the applicable airworthiness requirements:

14 October 2016



Date: 18 August 2021

TCDS No.: P.033 Issue: 03

2. EASA Certification Basis

2.1. Airworthiness Standards

CS-P amendment 1, dated 16 November 2006

2.2. Special Conditions (SC): None

2.3. Equivalent Safety Findings (ESF): None

2.4. Deviations: None

III. Technical Characteristics

1. Type Design Definition

The MTV-37 propeller consists of different design configurations. Each design configuration is by a main assembly drawing and associated parts list as per the following table:

Design Configuration	Assembly Drawing	Parts List		
MTV-37-1-E-()-() Constant Speed + Feather	P-1602-(x)	S-225-(x)		
MTV-37-1-E-C-F-R(G) Constant Speed + Feather	P-1581-(x)	S-221-(x)		
+ Reverse (System Garrett)				
MTV-37-1-E-C-F-R(M) Constant Speed + Feather	P-1601-(x)	S-224-(x)		
+ Reverse (System Mühlbauer)				
MTV-37-1-E-C-F-R(P) Constant Speed + Feather +	P-1528-(x)	S-215-(x)		
Reverse (System Pratt&Whitney)				
Note:				
There is one version of hub flange available, this being E: ARP-880, Type 1				



Date: 18 August 2021

TCDS No.: P.033 Issue: 03

2. Description

7-blade variable pitch propeller with a hydraulically operated blade pitch change mechanism providing the operation mode "Constant Speed", "Feather" and "Reverse". The hub is milled out of aluminium alloy. The blades have a laminated wood composite structure coated in with a composite fibre reinforced epoxy. The leading edge of the blade is equipped with an erosion protection device. Optional equipment includes spinner and ice protection.

3. Equipment

refer to MT-Propeller Service Bulletin No. 13 Spinner: refer to MT-Propeller Service Bulletin No. 14 Governor: Ice Protection: refer to MT-Propeller Service Bulletin No. 15

The equipment listed in SBs No.13, 14 and 15 is not included in the certified Type Design. Related propeller equipment must be approved as part of the aircraft installation regardless of manufacture.

4. Dimensions

Blade diameter: 160 cm to 242 cm

5. Weight

Depending on Propeller-Design Configuration

"Constant Speed": approx. 57 kg "Constant Speed, Feather": approx. 66 kg "Constant Speed, Reverse": approx. 69 kg "Constant Speed, Feather, Reverse": approx. 76 kg

6. Hub / Blade Combinations

For all design configurations listed under III.1 the following wooden blades are applicable:

```
-17, -24, -30, -32, -36, -39, -40, -53, -54, -56, -57, -59, -86, -100, -101, -105, -113, -114, -115, -117, -118, -
119, -130, -131, -301, -302, -351, -352, -353, -354, -355
```

7. Control System

Propeller governors as listed in MT-Propeller Service Bulletin No. 14.

8. Adaptation to Engine

Hub flange ARP-880, Type 1

9. Direction of Rotation

Direction of rotation (viewed in flight direction) as identified by a letter-code in the propeller designation (see VI.4.)



TE.CERT.00050-001 © European Union Aviation Safety Agency, 2021. All rights reserved. ISO9001 Certified. Page 6 of 10 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

Date: 18 August 2021

TCDS No.: P.033 Issue: 03

IV. Operating Limitations

1. Approved Installations

The suitability of a propeller for a given aircraft/engine combination must be demonstrated within the scope of the type certification of the aircraft. See also Note 5.

2. Maximum Take Off Power and Speed

Diameter (cm) Maximum Take Off Power (k\		Maximum Take Off Speed (rpm)
160-230	634	2000
160-242	533	1607

3. Maximum Continuous Power and Speed

Diameter (cm)	Maximum Continuous Power (kW)	Maximum Continuous Speed (rpm)
160-230	634	2000
160-242	533	1607

4. Propeller Pitch Angle

From -20° up to +86° measured at 75% radius station

V. Operating and Service Instructions

Manuals	
Operation and Installation Manual for hydraulically controlled variable pitch propeller	No. E-124 (*)
MTV-37-1-E-()-()	
Operation and Installation Manual for reversible hydraulically controlled variable pitch propeller; Reverse-Systems (M) MTV-37-1-E-C-F-R(M)	No. E-504 (*)
Operation and Installation Manual for reversible hydraulically controlled variable pitch propeller; Reverse-Systems (G), (P) MTV-37-1-E-C-F-R(G/P)	No. E-610 (*)

Date: 18 August 2021

TCDS No.: P.033 Issue: 03

_
No. E-124 (*)
No. E-504 (*)
No. E-610 (*)
No. E-220 (*)
No. E-519 (*)
No. E-680 (*)
No. E-1290 (*)
No. E-808 (*)
As published by MT-Propeller

^(*) latest revision of

VI. Notes

- 1. The EASA approved Airworthiness Limitations Section (ALS) of the Instructions for Continued Airworthiness is published in the applicable "Operation, Installation and Maintenance Manual" document, chapter 10.0 "Airworthiness Limitations Section". This ALS section is empty because no life limit is necessary for these models.
- 2. Some models of this propeller can incorporate a start pitch lock which may prevent propeller feathering below a given propeller speed.
- 3. The overhaul intervals recommended by the manufacturer are listed in MT-Propeller Service Bulletin No. 1.
- 4. Propeller designation system:

Hub

- 1 MT-Propeller Entwicklung GmbH
- 2 Variable pitch propeller



TE.CERT.00050-001 © European Union Aviation Safety Agency, 2021. All rights reserved. ISO9001 Certified. Page 8 of 10 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

Date: 18 August 2021

TCDS No.: P.033 Issue: 03

- 3 Identification of propeller type
- 4 Letter code for flange type:
 - -E = ARP-880, Type 1
- 5 Letter code for counterweights:
 - blank = no or small counterweights for pitch change forces to decrease pitch
 - C = counterweights for pitch change forces to increase pitch
- 6 Letter code for feather provision:
 - blank = no feather position possible
 - F = feather position allowed
- 7 Letter code for reverse provision:
 - blank = no reverse position possible
 - R = reverse position allowed
- 8 Letter code for reversing system:
 - M = System Mühlbauer
 - G = System Garrett
 - P = System Pratt & Whitney
- 9 Letter code for hub design changes:
 - small letter for changes which do not affect interchangeability
 - capital letter for changes which affect interchangeability

Blade

- 1 Letter code for position of pitch change pin:
 - blank = pin position for pitch change forces to decrease pitch
 - C = pin position for pitch change forces to increase pitch
 - CF = pin position to allow feather; pitch change forces to increase pitch
 - CR = pin position to allow reverse; pitch change forces to increase pitch
 - CFR = pin position to feather and reverse; pitch change forces to increase pitch
- 2 Letter code for direction of rotation and installation:
 - blank = right-hand tractor
 - RD = right-hand pusher
 - L = left-hand tractor
 - LD = left-hand pusher
- 3 Propeller diameter in cm
- 4 Identification of blade design
- 5 Letter code for blade design changes:
 - small letter for changes which do not affect interchangeability of blade set
 - capital letter for changes which affect interchangeability of blade set



TE.CERT.00050-001 © European Union Aviation Safety Agency, 2021. All rights reserved. ISO9001 Certified. Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.

Date: 18 August 2021

TCDS No.: P.033 Issue: 03

SECTION: ADMINISTRATIVE

I. Acronyms and Abbreviations

CFR Code of Federal Regulations

LBA Luftfahrt Bundesamt

II. Type Certificate Holder Record

As per I.2

III. Change Record

TCDS Issue	Date	Changes	TC Issue Date
Issue 01	09 December 2019	Initial issue	Initial Issue,
			09 December 2019
Issue 02	15 July 2020	Includes an new approved rating as per EASA	Initial Issue,
		Major Change Approval No. 10073792	09 December 2019
Issue 03	18 August 2021	Following propeller approval in accordance with	Initial Issue,
		Subpart D of CS-P, reference EASA approval No.	09 December 2019
		10077110, note 5. stating that the propeller is not	
		approved in accordance with Subpart D of CS-P	
		has been removed	

-END-



TE.CERT.00050-001 © European Union Aviation Safety Agency, 2021. All rights reserved. ISO9001 Certified. Page 10 of 10 Proprietary document. Copies are not controlled. Confirm revision status through the EASA-Internet/Intranet.