

EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10084642	REV. 2	321 PRECISION CONVERSIONS LLC	A321-211/-212/-213/-231/- 232	All requirements and conditions of AD 2016-0146R2 that are notspecifically referenced above remain fully applicable andmust becomplied with accordingly.This approval is applicable only to the Airbus A321 Manufacturer SerialNumber that have been converted to Freighter per EASA STC no. 10083629.Inspe	REV 1 to AMOC 10084642: AMOC applicability extended.REV 2 to AMOC 10084642: Typo correction.Paragraph (1) of AD 2016-0146R2 requires Structural inspections inaccordance with Airbus SBs in associated technical documentation ref.a), b) and c).321 Precision Conversions EASA STC 10083629 converts applic		a) Airbus Service Bulletin A320-53-1309, Revision 1 dated March 12, 2019b) Airbus Service Bulletin A320-53-1311, Revision 1 dated March 11, 2019c) Airbus Service Bulletin A320-53-1313, Revision 1 dated March 11, 2019d) 321 Precision Conversions Service Bulletin: A321PCF-53-0004 - FUSELAGE – FORWARD3	2016-0146R2	17/07/2024	Active
10086236		321 PRECISION CONVERSIONS LLC	A321-231/-232 A321-211/-212/-213	N/A	ATA 53 – Fuselage – Crossbeam Splicing at Frames 16 and 20 – InspectionEASA AD 2023-0150 requires inspection of certainfastener holes at FR16and FR20 on both LH and RH sides. Precision Conversions EASA STC10083629 does not affect the fastener holes in question, but theaccomplishment instructions re		Precision Conversions SB A321PCF-53-0005.	2023-0150	22/01/2025	Active
10073743		328 SUPPORT SERVICES GmbH	328-100	Applicable to Model 328-100 - all S/N	With CN-100519 EASA Major Change Approval No 10073701 an alternativeSpring Tab lever P/N 001A272A4020006 instead of P/ N001A272A4020004 aslisted in AD Section "Required action(s) and Compliance Time(s): No (5)"has been certified.The new P/N is considered also as terminating action to above listed AD			EASA 2008-0107	08/07/2020	Active
10076933		AEROCONSEIL	A320-232 A320-214 A320-212 A320-211 A319-114	None	ADT 406 ELT INSTALLATION ON A320 FAMILY.Following the Airworthiness Directive 2020-0103 published on the EASAwebsite dated on 07 May 2020 EASA, related to the Emergency LocatorTransmitter (ELT) having Part Number PN 01N65900 installed onlybyAirbus in production or as per Airbus Service Bulletin has		0242-05-A-ATI-F01-R00 or later revisions approved/ acceptedunder theEASA system.	2020-0103	15/07/2021	Active
10074765		AEROLOGIC Gmbh	777F	Serial Number(s): 36001, 36002, 36003, 36004, 36198, 36199,36200,36201, 37708, 37710, 66081, 66082, 66085, 66086, 66088, 66164, 66632,66814	Fuel - Centre Wing Tank Fuel Quantity Indicating System - Check / AFMRevisionAerologic GmbH submitted an application toEASA for an AMOC to ADUS-2020-11-11.AD US-2020-11-11 paragraphs (g)(1) through (6) define work instructionsto be accomplished by performing a Refuelling Station Door Cycling(RSDC)A		Flight Crew Operations Manual Bulletin for Aerologic GmbH ref. AGT-49dated July 16, 2019Flight Crew Operations Manual Volume 1 ref. 23TR dated June 15, 2020Maintenance TIP ref. 777 MT 28-031 R2 dated May 29, 2020	US-2020-11-11	06/11/2020	Active
10073510		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27602.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinitial and repeat inspections for overlapped skin panels as required byparagraph (g) mandated by Airworthiness Directive (AD) 2006-22-09applicable to Boeing model 747-4F6 airplane with S/N 27602.The subject aircraft was damaged		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Analysis (DTA) D20-0350-001 Rev 00,dated 06 Apr 2020.V1 Aerospace Repair Instructions R20-00350-001 Rev.04, dated 20 Mar2020.	AD 2006-22-09	16/06/2020	Active



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0073511		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27602.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinitial and repeat inspections and modifications of certain skin lapjoints as required by paragraph (g) mandated by Airworthiness Directive(AD) 2016-04-02 applicable to Boeing model 747-4F6 airplane with S/ N27602.The subject air		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Analysis (DTA) D20-0350-001 Rev 00,dated 06 Apr 2020.V1 Aerospace Repair Instructions R20-00350-001 Rev.04, dated 20 Mar2020.	AD 2016-04-02	16/06/2020	Active
0073512		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27602.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of thestructural inspections required by paragraph (i) mandated byAirworthiness Directive (AD) 2018-04-07 applicable to Boeing model747-4F6 airplane with S/N 27602.The subject aircraft was damaged in an incident involving ground suppo		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Analysis (DTA) D20-0350-001 Rev 00,dated 06 Apr 2020.V1 Aerospace Repair Instructions R20-00350-001 Rev.04, dated 20 Mar2020.	AD 2018-04-07	16/06/2020	Active
0074335		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 25700	Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinitial and repeat inspections and modifications of certain skin lapjoints as required by paragraph (g) mandated by Airworthiness Directive(AD) 2016-04-02 applicable to Boeing model 747-4H6 airplane with S/ N25700.The subject air		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Analysis (DTA) D20-0365-001 Rev 01,dated 10 Jul 2020.V1 Aerospace Repair Instructions R20-00365-001 Rev.03, dated 09 Apr2020.	2016-04-02	24/09/2020	Active
0074826		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinitial and repeat inspections and modifications of certain skin lapjoints as required by paragraph (g) mandated by Airworthiness Directive(AD) 2016-04-02 applicable to Boeing model 747-400 airplane with S/ N27899.The subject air		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Analysis (DTE) documentD20-00786-001 Rev 01, dated 29 October 2020V1 Aerospace Repair Instructions R20-00786-001 Rev 01 dated 21 July2020.	2016-04-02	13/11/2020	Active
0074832	REV. 1	AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of thestructural inspections mandated by paragraph (i) of AirworthinessDirective (AD) 2018-04-07 applicable to Boeing model 747-400 airplanewith S/N 27899.The subject aircraft had a gouge and dent in the RH fuselage skin at STA2040-20		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00786-001 Rev 01, dated 29 October 2020V1 Aerospace Repair Instructions R20-00786-001 Rev 01 dated 21 July2020.	2018-04-07	17/11/2020	Active
0075383		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899.	Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraphs (g)(1) and (j) ofAirworthiness Directive (AD) 2010-01-01, applicable to Boeing model747-400 series airplane with S/N 27899, for the repaired areas ofcracked RH and LH Main En		V1 Aerospace Repair R20-0745-001, Rev 08 - RH Main Entry Door N° 5Forward Reveal Frame,dated July 20,2020.V1 Aerospace Damage Tolerance Analysis (DTA) document D20-00745-001 Rev00, dated 23 Oct 2020******V1 Aerospace Repair R20-0762-001, Rev 08 - LH Main Entry Door N° 5Forward Reveal Frame,dated Ju	FAA 2010-01-01	12/01/2021	Active





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0075386		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899	Fuselage areas cracking inspections (2013-17-08 The Boeing Company: Docket No. FAA-2013-0097; Directorateldentifier 2011-NM-243-AD; Amendment 39-17572)Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraphs (o) ofAirworthiness Dire		V1 Aerospace Repair R20-0745-001, Rev 08 - RH Main Entry Door N° 5Forward Reveal Frame, dated July 20,2020.V1 AerospaceDamage Tolerance Analysis (DTA) document D20-00745-001 Rev00, dated 23 Oct 2020V1 Aerospace Repair R20-0762-001, Rev 08- LH Main Entry Door N° 5Forward Reveal Frame, dated July 22	EASA ADOPTED FAA 2013-17-08	13/01/2021	Active
10075388		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899	Inspections to detect cracks of all structure identified inBoeingdocument D6-35022 "Supplemental Structural InspectionDocument for Model747 Airplanes," Revision H, dated September 2013(2018-04-07 The Boeing Company: Amendment 39-19202; Docket No.FAA-2016-9067; Product Identifier 2016-NM-043-AD).Ai		V1 Aerospace Repair R20-0745-001, Rev 08 - RH Main Entry Door N° 5Forward Reveal Frame, dated July 20,2020.V1 AerospaceDamage Tolerance Analysis (DTA) document D20-00745-001 Rev00, dated 23 Oct 2020*******V1 Aerospace Repair R20-0762-001,Rev 08 - LH Main Entry Door N° 5Forward Reveal Frame, datedV	EASA ADOPTED FAA AD 2018-04-07	13/01/2021	Active
10075555		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400F with Serial Number (S/N) 27603.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of thecorrective actions required by paragraph (e) ofAirworthiness Directive (AD) 2005-03-01 applicable to Boeingmodel 747-400F airplane with S/ N27603.The subject aircraft features a repair of shim migration at theunderwing midspar f		V1 Aerospace Repair Instructions R21-00939-001, Rev. 01 dated 05 Jan2021.	EASA ADOPTED FAA 2005-03-01	02/02/2021	Active
10075642		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27066	Inspections to detect cracks of all structure identified inBoeingdocument D6-35022 "Supplemental Structural InspectionDocument for Model747 Airplanes," Revision H, dated September 2013(2018-04-07 The Boeing Company: Amendment 39-19202; Docket No.FAA-2016-9067; Product Identifier 2016-NM-043-AD).Ai		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00822-001 Rev 01, dated 31 Jan 2021V1 Aerospace Repair Instructions R20-00822-001 Rev 05 dated 19 Sep 2020.	EASA ADOPTED FAA AD 2018-04-07	16/02/2021	Active
10075643		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27066	Body Station 2598 Bulkhead Inner Chord and Web Inspection(2014-04-03 The Boeing Company: Amendment 39-17898; Docket No.FAA-2014-0432; Product Identifier 2014-NM-099-AD).Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinspections mandated by paragraph (i) of Airworthiness Dire		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00822-001 Rev 01, dated 31 Jan 2021V1 Aerospace Repair Instructions R20-00822-001 Rev 05 dated 19 Sep 2020.	EASA ADOPTED FAA AD 2014-14-03	16/02/2021	Active
10075762		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of thepost- inspection repair requirements mandated byparagraph (i) ofAirworthiness Directive (AD) 2001-11-06 applicable to Boeing model747-400 airplane with S/N 27898.During a C-check on the subject aircraft, a post-repair inspection		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00844-001 Rev 00, dated 08 Jan 2021V1 Aerospace Repair Instructions R20-00844-001 Rev 00 dated 03 Oct 2020.	EASA ADOPTED FAA 2001-11-06	02/03/2021	Active





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0075795		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898	Air Atlanta Icelandic has requested an AMOC to the accomplishment of therepair and inspection requirements mandated by paragraph (g) and (h)(2)of Airworthiness Directive (AD) 2020-15-04 applicable to Boeing model747-400 airplane with S/N 27898.During a C- check on the subject aircraft, a post- repairt		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00844-001 Rev 00, dated 08 Jan 2021V1 Aerospace Repair Instructions R20-00844-001 Rev 00 dated 03 Oct 2020.	EASA ADOPTED FAA AD 2001-15-14	05/03/2021	Active
0075832		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899.	Air Icelandic Atlanta has requested an AMOC to the Nose Wheel Well (NWW) post inspection repair requirements mandated byparagraph (h) ofAirworthiness Directive (AD) 2016-06-07 applicable to Boeing model747-400 series airplane with S/N 27899.During an inspection of the Nose Wheel Well pressure panels		V1 Aerospace Repair Instructions R21-00950-001 Rev.02 dated26 Jan 2021V1 Aerospace static substantiation S1-00950-001 Rev.00, dated 26 Jan2021.	EASA ADOPTED FAA AD 2016-06-07	11/03/2021	Active
0076272	REV. 1	AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899.	Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) ofAirworthiness Directive (AD) 2010-01-01, applicable to Boeing model747-400 series airplane with S/N 27899, for the repaired areas ofcracked LH Main Entry Door (MED) N°4D		Engineering Instruction EGAT E2010A52B-1529 Rev 01 dated 04Aug 2010V1 Aerospace Damage Tolerance Evaluation (DTE) document D20-00913-001Rev 00, dated 11 Feb 2021	EASA ADOPTED FAA AD 2010-01-01	22/04/2021	Active
0076278		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; Directorateldentifier 2008-NM-192-AD; Amendment 39-16157)Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) of		Engineering Instruction EGAT E2010A52B-1533 Rev 02 dated 04Aug 2010V1 Aerospace Damage Tolerance Evaluation (DTE) document D20-00914-001Rev 01, dated 11 Feb 2021	FAA 2010-01-01	22/04/2021	Active
0076283		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; Directorateldentifier 2008-NM-192-AD; Amendment 39-16157)Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) of		Engineering Instruction EGAT E2010A52B-1531 Rev 01 dated 04Aug 2010V1 Aerospace Damage Tolerance Evaluation (DTE) document D20-00915-001Rev 01, dated 19 Mar 2021	FAA 2010-01-01	22/04/2021	Active
0076284		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; Directorateldentifier 2008-NM-192-AD; Amendment 39-16157)Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) of		Engineering Instruction EGAT E2010A52B-1532 Rev 02 dated 04Aug 2010V1 Aerospace Damage Tolerance Evaluation (DTE) document D20-00916-001Rev 0, dated 10 Feb 2021	FAA 2010-01-01	22/04/2021	Active
0076285		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; DirectorateIdentifier 2008-NM-192-AD; Amendment 39-16157)Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) of		Engineering Instruction EGAT E2011A52B-1896 Rev 01 dated 09Oct 2011V1 Aerospace Damage Tolerance Evaluation (DTE) document D20-00917-001Rev 0, dated 11 Feb 2021	FAA 2010-01-01	22/04/2021	Active





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0076286		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899.	Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) ofAirworthiness Directive (AD) 2010-01-01, applicable to Boeing model747-400 series airplane with S/N 27899, for the repaired areas ofcracked RH Main Entry Door (MED) N°4C		Engineering Instruction EGAT E2011A52B-1885 Rev 01 dated 07Oct 2011V1 Aerospace Damage Tolerance Evaluation (DTE) document D20-00918-001Rev 0, dated 10 Feb 2021.	EASA ADOPTED FAA AD 2010-01-	22/04/2021	Active
0076287		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899.	Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) ofAirworthiness Directive (AD) 2010-01-01, applicable to Boeing model747-400 series airplane with S/N 27899, for the repaired areas ofcracked LH Main Entry Door (MED) N°5D		Engineering Instruction EGAT E2013A52B-444 Rev 01 dated 15 Mar 2013V1 Aerospace Damage Tolerance Evaluation (DTE) document D20-00920-001Rev 0, dated 11 Feb 2021.	EASA ADOPTED FAA AD 2010-01-01	22/04/2021	Active
0076301		AIR ATLANTA ICELANDIC	747-400 BCF	Boeing Model 747-400BCF with Serial Number (S/N) 24801	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; Directorateldentifier 2008-NM-192-AD; Amendment 39-16157)Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) of		V1 Aerospace Damage Tolerance Evaluation (DTE) document D20-00871-001Rev 0, dated 24 Jan 2021	FAA 2010-01-01	26/04/2021	Active
0076347		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F with Serial Number (S/N) 28263	Body Station 2598 Bulkhead Inner Chord and Web Inspection(2014-04-03 The Boeing Company: Amendment 39-17898; Docket No.FAA-2014-0432; Product Identifier 2014-NM-099-AD).Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinspections mandated by paragraph (i) of Airworthiness Dire	A20WE	Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00872-001 Rev 00, dated 05 Feb 2021Haeco Repair Scheme RSK-53-0042 Rev 00 dated 13 Jul 2018.	FAA 2014-14-03	28/04/2021	Active
0076348		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F with Serial Number (S/N) 28263	Inspections to detect cracks of all structure identified inBoeingdocument D6-35022 "Supplemental Structural InspectionDocument for Model747 Airplanes," Revision H, dated September 2013(2018-04-07 The Boeing Company: Amendment 39-19202; Docket No.FAA-2016-9067; Product Identifier 2016-NM-043-AD).Ai	A20WE	Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00872-001 Rev 00, dated 05 Feb 2021Haeco Repair Scheme RSK-53-0042 Rev 00 dated 13 Jul 2018.	FAA 2018-04-07	28/04/2021	Active
0076361		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F with Serial Number (S/N) 28263	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; Directorateldentifier 2008-NM-192-AD; Amendment 39-16157)Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) of	A20WE	V1 Aerospace Damage Tolerance Evaluation (DTE) document D20-00888-001Rev 0, dated 23 Feb 2021	FAA 2010-01-01	29/04/2021	Active
0076394		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F with Serial Number (S/N) 26563	Inspections to detect cracks of all structure identified inBoeingdocument D6-35022 "Supplemental Structural InspectionDocument for Model747 Airplanes," Revision H, dated September 2013(2018-04-07 The Boeing Company: Amendment 39-19202; Docket No.FAA-2016-9067; Product Identifier 2016-NM-043-AD).Ai	A20WE	Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00907-001 Rev 02, dated 13 Apr 2021	FAA 2018-04-07	04/05/2021	Active





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10076405		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F with Serial Number (S/N) 26563.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinitial and repeat inspections and modifications of certain skin lapjoints as required by paragraph (g) mandated by Airworthiness Directive(AD) 2016-04-02 applicable to Boeing model 747-400F airplane with S/ N26563.During a C-che	A20WE	Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00907-001 Rev 02, dated 13 Apr 2021.	EASA ADOPTED FAA AD 2016-04-02	05/05/2021	Active
10076425		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899.	Air Icelandic Atlanta has requested an AMOC to the Nose Wheel Well (NWW) inspections mandated by paragraph (g) of Airworthiness Directive (AD)2016-06-07 applicable to Boeing model 747-400 series airplane with S/N27899.During an inspection of the NWW pressure panels as per AD 2016-06-07 onBoeing model	A20WE	V1 Aerospace Damage Tolerance Evaluation (DTE) document D21-00950-001Rev 00, dated 30 Mar 2021V1 Aerospace Repair Instructions R21-00950-001 Rev.02 dated 26 Jan 2021.	EASA ADOPTED FAA AD 2016-06-07	10/05/2021	Active
10076511		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F with Serial Number (S/N) 28263.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of thestructural inspections mandated by paragraph (i) of AirworthinessDirective (AD) 2018-04-07 applicable to Boeing model 747-400F airplanefor the skin and structure repaired area of cargo door at STA 1790 ofS/N 28263.During service		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00884-001 Rev 00, dated 11 Apr 2021.	EASA ADOPTED FAA AD 2018-04-07	19/05/2021	Active
10076606		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F with Serial Number (S/N) 28263	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; DirectorateIdentifier 2008-NM-192-AD; Amendment 39-16157)Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) of		V1 Aerospace Damage Tolerance Evaluation (DTE) document D20-00885-001Rev 0, dated 13 Apr 2021	FAA 2010-01-01	02/06/2021	Active
10077051		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F with Serial Number (S/N) 28263	Inspections to detect cracks of all structure identified inBoeingdocument D6-35022 "Supplemental Structural InspectionDocument for Model747 Airplanes," Revision H, dated September 2013(2018-04-07 The Boeing Company: Amendment 39-19202; Docket No.FAA-2016-9067; Product Identifier 2016-NM-043-AD).Ai		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD20-00887-001 Rev 00, dated 05 Apr 2021Haeco Repair Scheme RSK-53-0042 Rev 00 date.	FAA 2018-04-07	03/08/2021	Active
0077052		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27602	Inspections to detect cracks of all structure identified inBoeingdocument D6-35022 "Supplemental Structural InspectionDocument for Model747 Airplanes," Revision H, dated September 2013(2018-04-07 The Boeing Company: Amendment 39-19202; Docket No.FAA-2016-9067; Product Identifier 2016-NM-043-AD).Ai		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) D21-00953-001 Rev00, dated 25 May 2021.V1 Aerospace Repair Instructions R21-00953-001 Rev.02, dated 30 Jan2021.	FAA 2018-04-07	03/08/2021	Active
10077056		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898	Nose Wheel Well sidewall panel web (2016-06-07 The Boeing Company: Amendment 39-18438; Docket No.FAA-2014-0774; Directorate Identifier 2013-NM-154-AD)Air Icelandic Atlanta has requested an AMOC to the Nose Wheel Well (NWW)inspections mandated by paragraph (g) of Airworthiness Directive (AD)2016-06-07		V1 Aerospace Damage Tolerance Evaluation (DTE) document D21-01036-001Rev 00 dated 09 Jun 2021.V1 Aerospace Repair Instructions R21-01036-001 Rev.02 dated 13 May 2021.	FAA 2016-06-07	03/08/2021	Active



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0077604		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898	Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraph (g) ofAirworthiness Directive (AD) 2010-01-01, applicable to Boeing model747-400 series airplane with S/N 27898, for the repaired areas ofcracked Main Entry Door (MED) N°4 Aft		Engineering Instruction EGAT E2011A52B1636 Rev 00 dated 02 Aug 2011V1 Aerospace Damage Tolerance Evaluation (DTE) document D21-01031-001Rev 00, dated 16 Sep 2021	FAA 2010-01-01	04/11/2021	Active
0078239		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 25700.This AMOC is only valid for 500 flight cycles (FC) or 12 months,whichever occurs first, from the time of repair (i.e., 15,633 FC or 16January 2023, whichever occurs first).	Air Atlanta Icelandic has requested an AMOC to the accomplishment of therepeat inspections and corrective actions as required by paragraph (g) mandated by Airworthiness Directive (AD) 2013-26-12 applicable to Boeingmodel 747-4H6 airplane with S/N 25700.The subject aircraft embodies a repair for wearb		V1 Aerospace Repair Instructions R22-01308-001 Rev.03, dated 16 Jan2022.V1 Aerospace Substantiation Static document S22-01308-001 Rev 00, dated16 Jan 2022.	EASA ADOPTED FAA AD 2013-26-12	01/02/2022	Active
10078329		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 28551.The test per Boeing SB 747-31-2332 Rev 1 must be performed after thede-modification of the aircraft.	Indicating/Warning – Overspeed Warning System - ModificationAir Atlanta Icelandic has requested an AMOC to the accomplishment of thede-modification of the resettable overspeed warning in accordance withBoeing Service Bulletin (SB) 747-31-2332 Rev 1 required for Boeing 747aircraft previously modified		Boeing Service Bulletin (SB) 747-31-2332 Rev 1	2009-0101	10/02/2022	Active
0079148		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 25700.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of therepeat inspections and corrective actions as required by paragraph (g) mandated by Airworthiness Directive (AD) 2013-26-12 applicable to Boeingmodel 747-4H6 airplane with S/N 25700.The subject aircraft embodies a repair for wearb		V1 Aerospace Repair Instructions R22-01308-001 Rev.03, dated 16 Jan2022.V1 Aerospace Damage Tolerance Evaluation (DTE) document D22-01038-001Rev 00, dated 04 March 2022.	EASA ADOPTED FAA AD 2013-26-12	02/05/2022	Active
0079303		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 BCF with Serial Number (S/N) 24801	Section 41, 42 and 46 Body Skin Lap Joint Inspections and Modifications (2016-04-02 The Boeing Company: Amendment 39-18396; Docket No.FAA-2015-2460; Directorate Identifier 2014-NM-163-AD.).Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinitial and repeat inspections of certai		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD22-01302-001 Rev 00 dated 10 April 2022.V1 Aerospace Repair Instructions R22-01302-001 Rev 03 dated 11 Jan 2022.	EASA ADOPTED FAA AD 2016-04-02	19/05/2022	Active
0079304		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 BCF with Serial Number (S/N) 24801.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of thedetailed visual inspections mandated by paragraph (g) of AirworthinessDirective (AD) 2009-19-05 applicable to Boeing model 747-400 convertedfreighter (BCF) airplane for the repaired area of fuselage aft of thenose radome (STA 14		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD22-01302-001 Rev 00 dated 10 April 2022.V1 Aerospace Repair Instructions R22-01302-001 Rev 03 dated 11 Jan 2022.	EASA ADOPTED FAA AD 2009-19-05	19/05/2022	Active
10079309		AIR ATLANTA ICELANDIC	747-400 BCF	Boeing Model 747-400 BCF with Serial Number (S/N) 24801	Inspections to detect cracks of all structure identified inBoeingdocument D6-35022 "Supplemental Structural InspectionDocument for Model747 Airplanes," Revision H, dated September 2013(2018-04-07 The Boeing Company: Amendment 39-19202; Docket No.FAA-2016-9067; Product Identifier 2016-NM-043-AD).Ai		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD22-01302-001 Rev 00 dated 10 April 2022.V1 Aerospace Repair Instructions R22-01302-001 Rev 03 dated 11 Jan 2022.	EASA ADOPTED FAA AD 2018-04-07	19/05/2022	Active





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10079321		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27899	Air Icelandic Atlanta has requested an AMOC to the corrective actionssubsequent to the Nose Wheel Well (NWW) top and side panels replacementmandated by paragraph (i) of Airworthiness Directive (AD) 2016-06-07applicable to Boeing model 747-400 series airplane S/N 27899 convertedto special freighter p		V1 Aerospace Damage Tolerance Evaluation (DTE) document D22-01311-001Rev 00 dated 28 Mar 2022.V1 Aerospace Repair Instructions R22-01311-001 Rev 00 dated 24 Jan 2022.	FAA 2016-06-07	20/05/2022	Active
10079352		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F with Serial Number (S/N) 30761.The test per Boeing SB 747-31-2332 Rev 1 must be performed after thede-modification of the aircraft.	Indicating/Warning – Overspeed Warning System - ModificationAir Atlanta Icelandic has requested an AMOC to the accomplishment of thede-modification of the resettable overspeed warning in accordance withBoeing Service Bulletin (SB) 747-31-2332 Rev 1 required for Boeing 747aircraft previously modified		Boeing Service Bulletin (SB) 747-31-2332 Rev 1	2009-0101	25/05/2022	Active
10079353		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400 with Serial Number (S/N) 30760.The test per Boeing SB 747-31-2332 Rev 1 must be performed after thede-modification of the aircraft.	Indicating/Warning – Overspeed Warning System - ModificationAir Atlanta Icelandic has requested an AMOC to the accomplishment of thede-modification of the resettable overspeed warning in accordance withBoeing Service Bulletin (SB) 747-31-2332 Rev 1 required for Boeing 747aircraft previously modified		Boeing Service Bulletin (SB) 747-31-2332 Rev 1	2009-0101	25/05/2022	Active
10079477		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 aircraft converted from Combi to special freighterconfiguration by STC EASA 10015911 with Serial Number (S/N) 27899.	Fuselage areas cracking inspections (2013-17-08 The Boeing Company: Docket No. FAA-2013-0097; Directorateldentifier 2011-NM-243-AD; Amendment 39-17572)Air Icelandic Atlanta has requested an AMOC to the accomplishment of theinitial and repeat inspections mandated by paragraphs (o) ofAirworthiness Dire		V1 Aerospace Repair R22-01317-001, Rev 02 - LH MED N° 5 Forward RevealFrame, BS2231, STR 17 dated 09 Feb 2022.V1 Aerospace Damage Tolerance Analysis (DTA) document D22-01317-001 Rev00, dated 17 Apr 2022.	FAA AD 2013-17-08	15/06/2022	Active
10079513		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 aircraft converted from Combi to special freighterconfiguration by STC EASA 10015911 with Serial Number (S/N) 27899.	Inspections to detect cracks of all structure identified inBoeingdocument D6-35022 "Supplemental Structural InspectionDocument for Model747 Airplanes," Revision H, dated September 2013(2018-04-07 The Boeing Company: Amendment 39-19202; Docket No.FAA-2016-9067; Product Identifier 2016-NM-043-AD).Ai		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD22-01341-001 Rev 00 dated 26 April 2022.V1 Aerospace Repair Instructions R22-01341-001 Rev 05 dated 24 Feb 2022.	FAA AD 2018-04-07	17/06/2022	Active
10079728		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) S/N 25700	Skin and Bear Strap Crack at the Aft Upper Corner of the Main Entry Door5 Cutout(2013-19-15 The Boeing Company: Amendment 39-17597; Docket No.FAA-2013-0211; Directorate Identifier2012-NM-230-AD.)Air Atlanta Icelandic has requested an AMOCto the corrective actionrequirements at the main entry doorc		V1 Aerospace Repair Instructions R22-01394-001 Rev 01 dated13 May 2022.V1 Aerospace Static Substantiation S22-01394-001 Rev 00 dated 09 June2022.	FAA AD 2013-19-15	15/07/2022	Active
10079936		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F with Serial Number (S/N) 26343.The test per Boeing SB 747-31-2332 Rev 1 must be performed after thede-modification of the aircraft.	Indicating/Warning – Overspeed Warning System - ModificationAir Atlanta Icelandic has requested an AMOC to the accomplishment of thede-modification of the resettable overspeed warning in accordance withBoeing Service Bulletin (SB) 747-31-2332 Rev 1 required for Boeing 747aircraft previously modified		Boeing Service Bulletin (SB) 747-31-2332 Rev 1	2009-0101	16/08/2022	Active



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10079937		AIR ATLANTA ICELANDIC	747-400 BCF	Boeing Model 747-400 BCF with Serial Number (S/N) 24975	Body Station 2598 Bulkhead inspection and modification(2010-14-07 The Boeing Company: Amendment 39-16352; Docket No.FAA-2008-0981; Product Identifier 2008-NM-073-AD).Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinspections mandated by paragraph (u) of Airworthiness Directi		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD22-01376-001 Rev 02, dated 23 June 2022	FAA 2010-14-07	16/08/2022	Active
10079941		AIR ATLANTA ICELANDIC	747-400 BCF	Boeing Model 747-400 BCF with Serial Number (S/N) 24975	Body Station 2598 Bulkhead Inner Chord and Web Inspection(2014-14-03 The Boeing Company: Amendment 39-17898; Docket No.FAA-2014-0432; Product Identifier 2014-NM-099-AD)Air Atlantalcelandic has requested an AMOC to the accomplishment of theinspections mandated by paragraph (i) of Airworthiness Direc		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD22-01376-001 Rev 02, dated 23 June 2022	FAA 2014-14-03	16/08/2022	Active
10079943		AIR ATLANTA ICELANDIC	747-400 BCF	Boeing Model 747-400 BCF with Serial Number (S/N) 24975	Body Station 2598 Bulkhead Upper Splice Fitting Inspection(2020-14-02 The Boeing Company: Amendment 39-21156; Docket No.FAA-2019-0990; Product Identifier 2019-NM-122-AD)Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinspections mandated by paragraph (g) of Airworthiness Dire		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD22-01376-001 Rev 02, dated 23 June 2022	FAA 2020-14-02	16/08/2022	Active
10080464		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; Directorateldentifier 2008-NM-192-AD; Amendment 39-16157)Air Icelandic Atlanta has requested an AMOC to the accomplishment of thecorrective actions and repeat inspections mandated by parag		V1 Aerospace Repair Instructions R22-01467-001 Rev 01 dated12 July 2022V1 Aerospace Damage Tolerance Evaluation (DTE) document D22-01467-001Rev 00 dated 30 Aug 2022	FAA 2010-01-01	26/10/2022	Active
10080477		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 aircraft (S/N) 25700 converted to IAI specialfreighter per EASA STC 10014423 (modified from PAX).	Repetitive inspection and modification of the upper deck tension tiesSTA 1120 through 1220(2016-05-12 The Boeing Company: Amendment 39-18430; Docket N°FAA-2015-2961; Directorate Identifier 2014-NM-145-AD)Air Atlanta Icelandic has requestedan AMOC to the inspections, modifications and post- modificati		Boeing Alert Service Bulletin 747-53A2507, revision 1, dated January 14,2010V1 Aerospace Repair R22-01402-001, R7, dated June 7, 2022V1 Aerospace Repair R22-01403-001, R4, dated June 14, 2022V1 Aerospace Repair R22-01404-001, R6, dated June 14, 2022	FAA AD 2016-05-12	28/10/2022	Active
10080479		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898	Rack and Fuselage Structure - Inspections of E-42 satellitecommunications (SATCOM) rack (99-14-04 BOEING: Amendment 39-11212. Docket 99-NM-45-AD)Air Icelandic Atlanta has requestedan AMOC to the accomplishment of therepeat inspections and corrective actions mandated by paragraph (a) and(b) of Airwor		V1 Aerospace Repair Instructions R22-01462-001 Rev 01 dated07 July 2022V1 Aerospace Damage Tolerance Evaluation (DTE) document D22-01462-001Rev 00 dated 22 Aug 2022	FAA AD 99-14-04	28/10/2022	Active



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080480		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898	Nose Wheel Well sidewall panel web (2016-06-07 The Boeing Company: Amendment 39-18438; Docket No.FAA-2014-0774; Directorate Identifier 2013-NM-154-AD)Air Icelandic Atlanta has requested an AMOC to the corrective actionssubsequent to the Nose Wheel Well (NWW) top and side panels replacementmandated by		V1 Aerospace Repair Instructions R22-01505-001 Rev 02 dated11 Sept2022.V1 Aerospace Static Analysis S22-01505-001 Rev 00 dated 28 Sept 2022.	FAA AD 2016-06-07	28/10/2022	Active
080488		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; DirectorateIdentifier 2008-NM-192-AD; Amendment 39-16157)Air IceIandic Atlanta has requested an AMOC to the accomplishment of thecorrective actions post repeat inspections mandated by para		V1 Aerospace Repair Instructions R22-01566-001 Rev 01 dated29 Sept 2022V1 Aerospace Static Analysis S22-01566-001 Rev 00 dated 13 Oct 2022.	FAA 2010-01-01	02/11/2022	Active
080983		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) S/N 25700.	Air Atlanta Icelandic has requested an AMOC to the post-repairinspections requirements at the main entry door 5 fuselage cutout at theupper aft corner as mandated by paragraph (g)of Airworthiness Directive(AD) 2013-19-15 affecting Boeing model 747-400 airplane serial number(S/N) 25700 converted toA		V1 Aerospace Repair Instructions R22-01394-001 Rev 01 dated13 May 2022.V1 Aerospace DTE D22-01394-001 Rev 00 dated 26 June 2022.	EASA ADOPTED FAA AD 2013-19-15	03/01/2023	Active
080991		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX)S/N 25700.	Air Atlanta Icelandic has requested an AMOC to the post-repairinspections requirements at the main entry door 5 (MED 5) fuselagecutout at the upper aft corner as mandated by paragraph (g) ofAirworthiness Directive (AD) 2010-01-01 affecting Boeing model 747-400airplane serial number (S/N) 25700 conv		V1 Aerospace Repair Instructions R22-01394-001 Rev 01 dated13 May 2022.V1 Aerospace DTE D22-01394-001 Rev 00 dated 26 June 2022.	EASA ADOPTED FAA AD 2010-01-01	05/01/2023	Active
080997		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) S/N 26343.	Wings – Front spar web inspections (2017-09-10 The Boeing Company: Amendment 39-18872; Docket No.FAA-2016-9394; Directorate Identifier 2016-NM-162-AD)Air Atlanta Icelandic has requested an AMOC to the repair requirementsat a fastener hole on the web of right hand (RH) wing front spar nearfront spar s		V1 Aerospace DTE D22-01633-001 Rev 00 dated 28 Nov 2022.	FAA AD 2017-09-10	06/01/2023	Active
0081087		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) S/N 26343.	Fuselage - Skin adjacent to the Drag Splice Fitting - Inspection(2001-22-04 Boeing: Amendment 39-12483; Docket 2000-NM-220-AD)Air Atlanta Icelandic has requested an AMOC to the repair and inspectionrequirements in the fuselage skin at the drag splice fitting at station (STA) 980 and stringer S-38R as		V1 Aerospace DTE D22-01632-001 Rev 00 dated 5 Dec 2022.	FAA AD 2001-22-04	20/01/2023	Active
0081093		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) S/N 26343.	Air Atlanta Icelandic has requested an AMOC to the inspectionrequirements in the fuselage skin at the drag splice fitting at station(STA) 980 and stringer S-38R as mandated by paragraph (g) ofAirworthiness Directive (AD) 2016-04-02affecting Boeing model 747-400airplane serial number (S/N) 26343 conv		V1 Aerospace DTE D22-01632-001 Rev 00 dated 5 Dec 2022.	EASA ADOPTED FAA AD 2016-04-02	20/01/2023	Active



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0081238	REV. 1	AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 aircraft (S/N) 25700 converted to IAI specialfreighter per EASA STC 10014423 (modified from PAX).	Air Atlanta Icelandic has requested an AMOC to the inspections, modifications and post-modification inspections of the upper decktension ties and shear webs mandated by Airworthiness Directive (AD) 2016-05-12 affecting Boeing model 747-400 series airplane converted toIAI special freighter.AD 2016-05-1		Boeing Alert Service Bulletin 747-53A2507, revision 1, dated January 14,2010V1 Aerospace Repair R22-01403-001, R4, dated June 14, 2022V1 Aerospace damage tolerance analysis document D22-01403-001 Rev 00dated 8 Dec 2022.	EASA ADOPTED FAA AD 2016-05-12	28/02/2023	Active
0081413		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 aircraft (S/N) 25700 converted to IAI specialfreighter per EASA STC 10014423 (modified from PAX).	Repetitive inspection and modification of the upper deck tension tiesSTA 1120 through 1220(2016-05-12 The Boeing Company: Amendment 39-18430; Docket N°FAA-2015-2961; Directorate Identifier 2014-NM-145-AD)Air Atlanta Icelandic has requestedan AMOC to the inspections,modifications and post- modificati		Boeing Alert Service Bulletin 747-53A2507, revision 1, dated January 14,2010V1 Aerospace Repair R22-01402-001, Rev 07 dated 7 June 2022V1 Aerospace damage tolerance analysis document D22-01402-001 Rev 00dated 14 Dec 2022V1 Aerospace RepairR22-01404-001 Rev 06 dated 14 June 2022V1 Aerospace damage t	FAA AD2016-05-12	07/03/2023	Active
0081416		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 aircraft converted from passengers to specialfreighter configuration by STC EASA 10014423 with Serial Number (S/N) 25700.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of thestructural inspections mandated by paragraph (i) of AirworthinessDirective (AD) 2018-04-07 applicable to Boeing model 747-400 aircraftconverted from passengers to special freighter configuration by STC EASA10014423 with Serial N		V1 Aerospace Repair R22-01402-001, Rev 07 dated 07 June 2022.V1 Aerospace damage tolerance analysis document D22-01402-001 Rev 00dated 14 Dec 2022.V1 Aerospace Repair R22-01404-001 Rev 06 dated 14 June 2022.V1 Aerospace damage tolerance analysis document D22-01404-001 Rev 00dated 16 Dec 2022.	EASA ADOPTED FAA AD 2018-04-07	08/03/2023	Active
0081442		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX)S/N 26343.	Air Atlanta Icelandic has requested an AMOC to the repetitiveinspections and repairs at the left hand size (LH) main entry door (MED)N°3 Body frame Station 1265 mandated by Airworthiness Directive (AD)2000-02-10 affecting Boeing model 747-400 aircraft serial number (S/N)26343 converted to IAI specia		Boeing Service Bulletin 747-53A2416, revision 3, dated April 25, 2002V1 Aerospace damage tolerance analysis document D22-01637-001 Rev 00dated 03 Jan 2023.	EASA ADOPTED FAA AD 2000-02-10	09/03/2023	Active
0081447		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX)S/N 26343.	Air Atlanta Icelandic has requested an AMOC to the post-repairinspections requirements at the left hand size (LH) main entry door(MED) N°3 Body frame Station 1265 as mandated by paragraph (g) ofAirworthiness Directive (AD) 2010-01-01 affecting Boeing model 747-400airplane serial number (S/N) 26343 c		V1 Aerospace damage tolerance analysis document D22-01637-001 Rev 00dated 03 Jan 2023.	EASA ADOPTED FAA AD 2010-01-01	10/03/2023	Active
0081579		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898.	Air Icelandic Atlanta has requested an AMOC to the accomplishment of thecorrective actions post inspections mandated byparagraph (j) ofAirworthiness Directive (AD) 2010-01-01, applicable to Boeing model747-400 series airplane with S/N 27898, for the repaired areas of righthand (RH) Main Entry DoorO		V1 Aerospace Repair Instructions R22-01566-001 Rev 01 dated29 Sept 2022V1 Aerospace damage tolerance analysis documentD22-01566-001 Rev 00dated 9 Feb 2023.	EASA ADOPTED FAA AD 2010-01-01	28/03/2023	Active





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10081595		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Air Icelandic Atlanta has requested an AMOC to the investigative andcorrective actions at door 5 right ceiling stowage boxes as per BoeingSpecial Attention Requirements Bulletin 747-25-3726 RB, dated January 6,2022 and as mandated by paragraph (g) of Airworthiness Directive (AD)2022-25-04 affectingI			EASA ADOPTED FAA AD 2022-25-04	28/03/2023	Active
10081598		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898	Nose Wheel Well sidewall panel web (2016-06-07 The Boeing Company: Amendment 39-18438; Docket No.FAA-2014-0774; Directorate Identifier 2013-NM-154-AD)Air Icelandic Atlanta has requested an AMOC to the corrective actionssubsequent to the Nose Wheel Well (NWW) top and side panels replacementmandated by		V1 Aerospace Repair Instructions R22-01505-001 Rev 02 dated11 Sept2022.V1 Aerospace damage tolerance analysis documentD22-01505-001 Rev 00dated 7 Feb 2023.	FAA 2016-06-07	29/03/2023	Active
10081828		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX)S/N 26343.	Air Atlanta Icelandic has requested an AMOC to the repair and repetitiveinspections of the horizontal stabilizer left hand upper skin, asmandated by Airworthiness Directive (AD) 2006-16-10 paragraph (f), (l)and (m) affecting Boeing model 747-400 series airplane converted toIsrael Aerospace Industrie		- EGAT Engineering Instructions A2019A55M-1283, Revision 1,dated April6, 2019- V1 Aerospace damage tolerance analysis document D22-01650-001 Rev 00dated 09 Feb 2023	EASA ADOPTED FAA AD 2006-10-16	04/05/2023	Active
10081844		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX)S/N 26343.	Air Atlanta Icelandic has requested an AMOC to the inspectionrequirements for the horizontal stabilizer left hand upperskin, asmandated by Airworthiness Directive (AD) 2016-25-05paragraph (g) affecting Boeing model 747-400 series airplaneconverted to IsraelAerospace Industries (IAI) special freigh		- EGAT Engineering Instructions A2019A55M-1283, Revision 1,dated April6, 2019- V1 Aerospace damage tolerance analysis document D22-01650-001 Rev 00dated 09 Feb 2023.	EASA ADOPTED FAA AD 2016-25-05	05/05/2023	Active
10081889		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX)S/N 26343.	Air Atlanta Icelandic has requested an AMOC to the inspectionrequirements for the horizontal stabilizer left hand upperskin, asmandated by Airworthiness Directive (AD) 2010-05-03paragraph (g) affecting Boeing model 747-400 series airplaneconverted to IsraelAerospace Industries (IAI) special freigh		- EGAT Engineering Instructions A2019A55M-1283, Revision 1,dated April6, 2019- V1 Aerospace damage tolerance analysis document D22-01650-001 Rev 00dated 09 Feb 2023.	EASA ADOPTED FAA AD 2010-05-03	09/05/2023	Active
10081894		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX)S/N 26343.	Air Atlanta Icelandic has requested an AMOC to the inspectionrequirements mandated by Airworthiness Directive (AD) 2018-04-07paragraph (i) affecting Boeing model 747-400 series airplane convertedto Israel Aerospace Industries (IAI) specialfreighter.During accomplishment of inspections per Boeing Se		- EGAT Engineering Instructions A2019A55M-1283, Revision 1,dated April6, 2019- V1 Aerospace damage tolerance analysis document D22-01650-001 Rev 00dated 09 Feb 2023.	EASA ADOPTED FAA AD 2018-04-07	09/05/2023	Active
10081933		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) S/N 26343.	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; DirectorateIdentifier 2008-NM-192-AD; Amendment 39-16157)Air Atlanta Icelandic has requested an AMOC to the inspection andcorrective action requirements at various fuseIage internal struct		V1 Aerospace damage tolerance analysis document D22-01638-001 Rev 00dated 04 Jan 2023	FAA 2010-01-01	15/05/2023	Active



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0081938		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX)S/N 26343.	Air Atlanta Icelandic has requested an AMOC to the inspectionrequirements at various fuselage internal structure locations asmandated by paragraph (i) of Airworthiness Directive (AD) 2018-04-07affecting Boeing model 747-400 airplane serialnumber (S/N) 26343converted to IAI special freighter.Duringf		V1 Aerospace damage tolerance analysis document D22-01638-001 Rev 00dated 04 Jan 2023.	EASA ADOPTED FAA AD 2018-04-07	15/05/2023	Active
0082035		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) S/N 26343.	Repetitive inspection and modification of the upper deck tension tiesSTA 1120 through 1220(2016-05-12 The Boeing Company: Amendment 39-18430; Docket N°FAA-2015-2961; Directorate Identifier 2014-NM-145-AD)Air Atlanta Icelandic has requestedan AMOC to the modifications andpost-modification inspection		V1 Aerospace damage tolerance analysis document D22-01651-001 Rev 00dated 24 Feb 2023.	FAA 2016-05-12	02/06/2023	Active
0082151		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10015911 (modified from Combi)S/N 27898.	Air Atlanta Icelandic has requested an AMOC to the inspectionrequirements for the horizontal stabilizer left hand upperskin, asmandated by Airworthiness Directive (AD) 2010-05-03paragraph (g) affecting Boeing model 747-400 series airplaneconverted to IsraelAerospace Industries (IAI) special freigh		- V1 Aerospace Repair Instructions, R22-01503-001, Revision03, datedAugust 19, 2022 V1 Aerospace damage tolerance analysis document D22-01503-001 Rev 00dated 27 Mar 2023.	EASA ADOPTED FAA AD 2010-05-03	15/06/2023	Active
0082174		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10015911 (modified from Combi) S/N 27898.	Inspections to detect cracks of all structure identified inBoeingdocument D6-35022 "Supplemental Structural InspectionDocument for Model747 Airplanes," Revision H, dated September 2013(2018-04-07 The Boeing Company: Amendment 39-19202; Docket No.FAA-2016-9067; Product Identifier 2016-NM-043-AD).Ai		V1 Aerospace Repair Instructions, R22-01503-001, Revision 03, datedAugust 19, 2022.V1 Aerospace damage tolerance analysis document D22-01503-001 Rev 00dated 27 Mar 2023	FAA 2018-04-07	19/06/2023	Active
0082187		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10015911 (modified from Combi)S/N 27898.	Air Atlanta Icelandic has requested an AMOC to the inspectionrequirements for the horizontal stabilizer left hand upperskin, asmandated by Airworthiness Directive (AD) 2016-25-05paragraph (g) affecting Boeing model 747-400 series airplaneconverted to IsraelAerospace Industries (IAI) special freigh		- V1 Aerospace Repair Instructions, R22-01503-001, Revision03, datedAugust 19, 2022 V1 Aerospace damage tolerance analysis document D22-01503-001 Rev 00dated 27 Mar 2023.	EASA ADOPTED FAA AD 2016-25-05	20/06/2023	Active
0082216		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10015911 (modified from Combi)S/N 27898.	Air Atlanta Icelandic has requested an AMOC to the inspectionrequirements mandated by Airworthiness Directive (AD) 2018-04-07paragraph (i) affecting Boeing model 747-400 series airplane convertedto Israel Aerospace Industries (IAI) specialfreighter.During C-check inspections on aircraft serial numb		- V1 Aerospace Repair Instructions, R22-01503-001, Revision03, datedAugust 19, 2022 V1 Aerospace damage tolerance analysis document D22-01477-001 Rev 00dated 29 Mar 2023.	EASA ADOPTED FAA AD 2018-04-07	23/06/2023	Active
0082361		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10015911 (modified from Combi) S/N 27898.	Main Frame – Inspections based on Fatigue Test Results(2010-01-01 The Boeing Company: Docket No. FAA-2009-0655; DirectorateIdentifier 2008-NM-192-AD; Amendment 39-16157)Air Atlanta Icelandic has requested an AMOC to the inspectionrequirements mandated by Airworthiness Directive (AD) 2010-01-01paragr		V1 Aerospace Repair Instructions, R22-01503-001, Revision 03, datedAugust 19, 2022.V1 Aerospace damage tolerance analysis document D22-01477-001 Rev 00dated 29 Mar 2023	FAA 2010-01-01	10/07/2023	Active



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10082673		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F S/ N 30761.	Air Atlanta Icelandic has requested an AMOC to the inspectionrequirements for the horizontal stabilizer right hand upper skin, asmandated by Airworthiness Directive (AD) 2016-25-05 paragraph (g) affecting Boeing model 747-400F with serial number (S/N) 30761.During accomplishment of the inspections re		V1 Aerospace damage tolerance analysis document D22-01922-001 Rev 00dated 19 Jul 2023.	EASA ADOPTED FAA AD 2016-25-05	28/08/2023	Active
0083608		AIR ATLANTA ICELANDIC	747-400	Boeing Model 747-400 with Serial Number (S/N) 25700 25700 converted toIAI special freighter per EASA STC 10014423 (modified from PAX).	Fuselage skin (2013-26-12 The Boeing Company: Amendment 39-17721; DocketNo. FAA-2013-0540; Directorate Identifier 2012-NM-185-AD)Air Atlanta Icelandic has requested an AMOC to theaccomplishment of thecorrective actions as required by paragraph (g) mandated byAirworthiness Directive (AD) 2013-26-127		V1 Aerospace Repair Instructions R23-02068-001 Rev.03, dated 29 Nov. 2023.V1 Aerospace Substantiation Static document S23-02068-001 Rev 00 dated02 Dec 2023.	FAA 2013-26-12	20/12/2023	Active
10084872		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400F MSN 30760	Repetitive inspections of the front spar shear tie and intercostal lugfitting at certain locations for any cracking. (2021-17-06 The Boeing Company: Amendment 39-21689; Docket No.FAA-2021-0376; Project Identifier AD-2021-00062-T.)Air Atlanta Icelandic has requested an AMOC to the repetitive detaileda		Document V1 Aerospace Repair Instructions R24-02274-001 Rev02 dated 23June 2024Static substantiation document V1 Aerospace S24-02274-001 Rev 00 dated23 June 2024	FAA 2021-17-06	30/07/2024	Active
0086058		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400 with Serial Number (S/N) 27503.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of thestructural inspections mandated by paragraph (i) of AirworthinessDirective (AD) 2018-04-07 applicable to Boeing model 747-400F airplanewithS/N 27503.During inspection on the subject aircraft, a crack was found on the AFTcargo do		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD24-02305-001 Rev 00, dated 8 Nov 2024Repair Scheme RS-19827, Revision 01, dated October 17, 2012.	EASA ADOPTED FAA AD 2018-04-07	17/12/2024	Active
0086068		AIR ATLANTA ICELANDIC	747-400F	Boeing Model 747-400 with Serial Number (S/N) 27503.	Air Atlanta Icelandic has requested an AMOC to the accomplishment of theinspections mandated by paragraph (a) and (c) of Airworthiness Directive(AD) 2003-10-06 applicable to Boeing model 747-400F airplane with S/N27503.During inspection onthe subject aircraft, a crack was found on the AFTcargo door		Section 5.0 Instructions for Continued Airworthiness (ICA) of V1Aerospace document Damage Tolerance Evaluation (DTE) documentD24-02305-001 Rev 00, dated 8 Nov 2024Repair Scheme RS-19827, Revision 01, dated October 17, 2012.	EASA ADOPTED FAA AD 2003-10-06	17/12/2024	Active
10076725		AIRBUS DEFENCE AND SPACE S.A.	A330-200, A330-243	A330-MRTT STC 10069541:0747 0892 0898 0951 0969 0980 0996 1036 1080 1183 1186 1235 1379 12501478 1516 1667 1735 1762 1787 1799 1808 1822 1848 1857 1883 1886 18911911 1916 1919 1942 1945 1960 19651968 1982 1989 1998 2008 2011A330- FSTA STC1 10034690 and STC2 10035945:1033 1046 1248 1275 1312 1334 1363	For the Serial Numbers listed in the Limitations and Conditionsparagraph of this AMOC (A330-MRTT, aircrafts modified bythe STCsincluded in the STC Approved Model List (10069541) and for A330-FSTA,aircrafts modified by STC2 10035945 and STC1 10034690), the AirbusDefence and Space Service Bulletin A3			2020-0273	29/06/2021	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
0077992		AIRBUS DEFENCE AND SPACE S.A.	A330-203		At the time of LHT STC 10067230 installation in A330-200 MSN 0898, EASAAD 2010- 0135 (and AD 2019-0155) has been assessed by LHT DesignOrganization as "Not applicable for zone 842" (Passenger/Crew/ EmergencyExit Door 2R), since this door isdeactivated<(>,<)> and the associatedsystem components addre		LHT SIL-00103 dated 21 January 2021.	2019-0155	16/12/2021	Active
0083771		AIRBUS DEFENCE AND SPACE S.A.	A330-243 A330-203 A330-202	Limited to A330-202/-203/-243 aircraft converted to MRTT/FSTA throughany of the following Airbus Defence and Space EASASTC: 10029272 (RAAF,RSAF and UAE), 10063084 (GOS), 10064192(FAF), 10064600 (RoKAF), 10064600 (RoKAF), 10069256 (MMF), 10035945 (FSTA 2) at any revision	ATA 28 – Fuel – Inner Fuel Tanks – Leak TestAirbus S.A.S. Service Bulletin (SB) A330-28-3141 requires performing aleak test to check the integrity of the Rib 3 aft boundary. It has beenmandated by EASA AD 2023-0052. The required leak test is applicable tothe A330 MRTT/FSTA fleet but cannot be direct		Airbus Defence and Space SB A330MRTT-28-0010 REV 00 dated 2023-12-22, orlater approved revisions of this document	2023-0052	26/01/2024	Active
0085394		AIRBUS DEFENCE AND SPACE S.A.	A330-202/-203/-243	This approval shall remain valid as long as the referenced AD is notcancelled or superseded.	EASA AD 2023-0012 mandates Airbus S.A.S. Alert Operators Transmission (AOT) A11L001-22. This AOT requires inspecting theleft wing surge tanklower surface to check the proper installation of warning placards P/ND1121313320000, and if missing, to install them.Airbus Defence and Space (ADS) has issuedA		Airbus Defence and Space SB A330MRTT-11-0029 REV 00 dated 01- AUG-2024The use of later approved revisions of this document is acceptable forcompliance with the requirements of thisAMOC.	2023-0012	27/09/2024	Active
0072585		AIRBUS HELICOPTERS	AS 332 L	This AMOC is valid for the helicopter AS 332 L, Serial Number 2113provided that this helicopter stays in a configuration of partialremoval of the de- icing system as defined above.Should the helicopterbe otherwise modified, it will have tocomply with altogetherrequirement of the AD 2018-0142R1.	Partial removal of the de-icing system, as accomplished on thehelicopter AS332L, S/N 2113 in accordance with Airbus Helicoptersinstructions described below (see Associated Technical Documentation), is adequate Alternative Method of Compliance to the AD 2018-0142R1 toprevent that helicopter to suffer	Applicability = Model: AS 332 L; Serial No.: 2113	Airbus Helicopters doc. ETG478/17 issue EAirbus Helicoptersdoc. Technical Agreement AS332-02-001C	2018-0142R1	25/02/2020	Active
0076125		AIRBUS HELICOPTERS	AS 332 L1 AS 332 C, AS 332 L	Limited to Serial Numbers: 2332, 2351, 2053, 2067, 2132, 2151, 2218<(>,<)> 9002, 2001, 2347, 9007, 2155, 2221, 9004<(>,<)> 9003, 2145, 2143, 2153 and 2341.	As Alternative Method of Compliance (AMOC) to the requirements of AD2020-0022R2, Paragraph (2):For Group 1 helicopters:Within 300 FH or within 31 August 2021,whichever occurs first after 21 February 2020 [the effective date of theoriginalissue of this AD], modify the helicopter by installing FFMPa		None	2020-0022R2	07/04/2021	Active
0076393		AIRBUS HELICOPTERS	EC120B		ATA 32 – Landing Gear – Skid Tubes – InspectionParagraph (5) of EASA AD 2020-0247 is replaced by the following:(5) Replacing each affected part on a helicopter under the conditions oftable 1 is an acceptable alternative method to comply with therequirements of paragraph (4) of this AD for that helic	EC120 B	ETIFF n°2021-0027	2020-0247	04/05/2021	Active





EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
0076963		AIRBUS HELICOPTERS	EC 225 LP	This AMOC is limited to the EC225 LP helicopter S/Ns 2662, 2749, 2753,2759, 2767, 2792, 2927 and 2973.	As Alternative Method of Compliance (AMOC) to requirement (6) of EASA AD2019-0074 mandating the replacement of the MainRotor (M/R) rotatingswashplate Part Number (P/N) 332A31-3074-00 or P/N 332A31-3074-01 beforeit exceeds 12 years since new (date of manufacture), accomplish thefollowing action:-,,i		Airbus Helicopters (AH) Emergency Alert Service Bulletin (EASB) ref.EC225-05A051 Revision 2, dated 26 February 2019;AH document ref. ETMD 2021_0330 issue A, released on 20 July 2021;AH document ref. ETMD N°2018-0215 issue B, released on 29June 2018;or later approved revisions of these documents.	2019-0074	20/07/2021	Active
0077146		AIRBUS HELICOPTERS	EC 225 LP	This AMOC is limited to the EC225 LP helicopter S/Ns 2914 and 2986.	ATA 25 – Equipment / Furnishings – Hoist – Test / ReplacementAs Alternative Method of Compliance (AMOC) to requirement (7), Table 4,of EASA AD 2015-0226R5, Interval applicable to All Affected Helicopters, accomplish the following action:-atintervals of 36 months (with a margin of 4 months), or 20000		EASA Major Change Approval 10073295 dated 18 May 2020;Airbus Helicopters ALS NR 013 dated 08 March 2021;or later approved revisions of these documents.	2015-0226R5	27/08/2021	Active
0077724		AIRBUS HELICOPTERS	EC 120 B		Paragraph (5) of EASA AD 2020-0247 is replaced by the following:(5) Replacing each affected part on a helicopter under the conditions oftable 1 is an acceptable alternative methodto comply with therequirements of paragraph (4) of this AD for that helicopter providedthat:-upon installation an intern		ETIFF n°2021-0027	2020-0247	19/11/2021	Active
0077729		AIRBUS HELICOPTERS	EC 225 LP	This AMOC is limited to the EC225 LP helicopter S/Ns 2741 and 2752.	As Alternative Method of Compliance (AMOC) to requirement (7), Table 4, of EASA AD 2015-0226R5, Interval applicable to All AffectedHelicopters, accomplish the following action:- at intervals of 36 months (with a margin of 4 months), or 2000 hoistoperating cycles (HC), whichever occurs first, perform		EASA Major Change Approval 10073295 dated 18 May 2020;Airbus Helicopters ALS NR 013 dated 08 March 2021;or later approved revisions of these documents.	2015-0226R5	22/11/2021	Active
0077949		AIRBUS HELICOPTERS	EC 155 B1	This AMOC is limited to the EC 155 B1 helicopter S/N 6915 only	As Alternative Method of Compliance (AMOC) to requirement (5) and Table3 of EASA AD 2018-0263, accomplish the following actions:for Group 1 helicopter equipped with affected MGB that has at least onetype Z planet gear installed, which accumulates 1.800 FH or more, within660FH after 16 March 2018 [t		Airbus Helicopters ASB EC15505A034 Revision 5Airbus Helicopters SB EC155-63-016 Revision 3	2018-0263	15/12/2021	Active
0081851		AIRBUS HELICOPTERS	EC 225 LP	Inspections as required by this AMOC shall only be accomplished byqualified personnel for crack detection by Eddy currents.All other requirements of EASA AD 2023-0042 shall be complied with, asapplicable.	As Alternative Method of Compliance (AMOC) to the requirement ofParagraph (7) of EASA AD 2023-0042, accomplish the following actions:(7.1),,When the affected part (as defined in AD 2023-0042) reaches 13years since new (date of manufacture), and thereafter at intervals notto exceed 100 FH, inspect ea		Airbus Helicopters Work Card 20-90-03-108 (MTC) Airbus Helicopters EC225 Emergency ASB 05A051 at Revision 6 dated 10February 2023, or later approved revisionsAirbus Helicopters document ref. N_ETIR_2023_0025.	AD 2023-0042	05/05/2023	Active
0082168	REV. 1	AIRBUS HELICOPTERS	EC 175 B	Applicable to Serial Number 5009.	As Alternative Means of Compliance (AMOC) to the requirements of AD2021-0178R2, Paragraph (1)Before next flight after 25 June 2019 [the effective date of EASAEmergency AD 2019-0148-E], and, thereafter, at intervals not to exceed10 flight hours (FH), inspect the horizontal stabilizer main spar inacco		AH Letter 2023-ETJA-00.	EASA 2021-0178R2	20/06/2023	Active



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0083875		AIRBUS HELICOPTERS	EC 130 B4, EC 130 T2	This AMOC is limited to:- EC 130 B4 rotorcraft MSN: 3842, 4111, 4272, 7325, 7413, 7518- EC 130 T2 rotorcraft MSN: 8122,8162, 7790.	All inspection requirements and intervals as mentioned in EASA AD2018-0104R2 remain applicable, except for paragraph (2) of this AD. Forparagraph (2) of the AD, the new interval is now 10 FH or 150 slingcycles, whichever occurs first.For the modification, as required by paragraph (10) or (11) of the			2018-0104 R2	08/02/2024	Active
0085699		AIRBUS HELICOPTERS	EC 225 LP	Inspections as required by this AMOC shall only be accomplished byqualified personnel for crack detection by Eddy currents.All other requirements of EASA AD 2024-0205 shall be complied with, asapplicable.	ATA 62 – Main Rotor – Rotating Swashplate Yokes – Inspection / Restoration / Service Life LimitAs Alternative Method of Compliance (AMOC) to the requirement ofParagraph (7) of EASAAD 2024-0205, accomplish the following action:(7.1) When the affected part (as defined in AD 2024-0205) reaches 13yearsn		AH EC225 EASB 05A051 Revision 7, dated 08 October 2024;or later approved revisions of this document.	2024-0205	30/10/2024	Active
0077047		AIRBUS MILITARY SOCIEDAD LIMITADAAMSL	A400M-180	None	ATA 71 – Power Plant – Engine Vibration Isolation System – Inspection / ReplacementEASA AD 2021-0077 requests to performrepetitive inspections of theforward lower Engine VibrationIsolation Systems (EVIS). The objectiveof these inspectionsis to detect a premature EVIS housing bolt failurethat, if s		Airbus Defence and Space (D<(>&<)>S) All Operators Transmission (AOT)AOT- A400M-71-0001 Revision 1	2021-0077	02/08/2021	Active
0072467		AIRBUS S.A.S.	A340-541/-542/-642/-643 A340-311/-312/-313 A340-211/-212/-213 A330-341/-342/-343 A330-321/-322/-323 A330-301/-302/-303 A330-223/-223F/-243/-243 F A330-201/-202/-203	None	Paragraphs 1 and 2 of the AD 2017-0091R2 read:(1) For Group1 aeroplanes: Within 12 months after the effective date of this AD, update the software of the affected processor, or replace itwith a TPA-100B processor P/N 940-0351-005, in accordance with theinstructions of the applicable Airbus SB.(2) An			2017-0091R2	11/02/2020	Active
0072500		AIRBUS S.A.S.	A320-271N A320-251N, A320-253N		ATA – Aircraft Flight Manual Section Limitations – AmendmentThis AD mandates Airbus Aircraft Flight Manual (AFM) Temporary Revision(TR) 773 issue 1, TR 774 issue 1, TR 775 issue 1and TR 776 issue 1, asapplicable, introducing limitations within the Centre of Gravityenvelope to mitigate the reduced e		Airbus Major Modification 163923Airbus Flight Operations Transmission 999.0059/19	2019-0189	11/02/2020	Active
0072604		AIRBUS S.A.S.	A380-841	None	ATA 54 – Nacelles / Pylons – Inboard and Outboard Pylon Drain System – Inspection / ModificationAD 2020-0025 mandates repetitive inspections as per SB A380-54-8073until embodiment of the applicable Modification SBs (MSB) whichconstitute terminating action of the Inspection SB A380-54-8073.For Pylon 3		RDAS ref. 80662998/003/2019	2020-0025	28/02/2020	Active
0072605	REV. 1	AIRBUS S.A.S.	A320-251N/-253N/-271N	This AMOC Revision 1 approval supersedes:- the AMOC approval 10072500, dated 11 February 2020, and- the AMOC 10072605 Original issue, dated 28 February 2020.	ATA – Aircraft Flight Manual Section Limitations – AmendmentThe AD 2019-0189 mandates Airbus Aircraft Flight Manual (AFM) TemporaryRevision (TR) 773 issue 1, TR 774 issue 1, TR 775 issue 1 and TR 776issue 1, as applicable, introducing limitations within the Centre ofGravity envelope to mitigate theA		Airbus Major Modification 163923Airbus Flight Operations Transmission 999.0059/19Airbus Flight Operations Transmission 999.0012/20Airbus Service Bulletin A320-27-1284 (at any revision)	AD-2019-0189	12/03/2020	Active



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10073217		AIRBUS S.A.S.	A320-214, A320-232	None	Background:EASA AD 2019-0279R1 was issued to remove a potential unsafe conditiondue to insufficient clearance between the cargo fire extinguishingpipes.MSNs 4170 and 5255 are not impacted by the issue described in EASA AD2019-0279R1 being modified according to EASA STC 10031840 (MSN 4170) &EASA STCr		- Technical Note AJEY-2020-0011/SD Rev A- Airbus SB A320-26-1109 Revision 1- MOD CJ0648: Cargo Compartment Fire Extinguishers Relocation- EASA STC 10031840 (MSN 4170)- EASA STC 10050094 (MSN 5255)- TDDS26001: Technical Design Directive regarding ATA26 on A320- AIS SB 601903-25-000101-002	2019-0279R1	08/05/2020	Active
10073857		AIRBUS S.A.S.	A321-213/-231/-232 A321-111/-112/-131/-211/- 212 A320-231/-232/-233 A320-211/-212/-214/-215/- 216 A319-131/-132/-133 A319-111/-112/-113/-114/- 115 A318-111/-112/-121/-122	None	EASA AD 2017-0237 was issued to detect an eventual engagement of theSecondary Load Path (SLP) of the Trimmable Horizontal StabilizerActuator (THSA).Among the various actions required by the EASA AD 2017-0237, it has beenidentified that some tasks referred through the mandated technicalpublications h		Technical Adaptation (TA) 80783356/007/2020 dated April 30th, 2020.	2017-0237	22/07/2020	Active
10073967		AIRBUS S.A.S.	A321-251N/-253N/-271N A321-231/-232 A321-211/-212/-213 A321-111/-112/-131 A320-251N/-271N A320-231/-232/-233 A320-211/-212/-214/-215/- 216 A319-131/-132/-133 A319-111/-112/-113/-114/- 115 A318-111/-112/-1231/-122	None	ATA 34 – Navigation – Traffic Collision Avoidance System Processor – Modification (Software Update) / ReplacementEASA AD2017-0091R2 was issued to address the number of false TCASresolution advisories (RA) by modifying the traffic collisionavoidancesystem (TCAS) processor through an upgrade of the s		-Airbus SB A320-34-1760 dated September 09, 2019<(>,<)> forA320 CEOmodels-Airbus SB A320-34-1761 dated September 09, 2019, for A320 NEO modelsAirbus modification 166196	2017-0091R2	05/08/2020	Active
10074245		AIRBUS S.A.S.	A380-841/-842/-861	None	AD 2020-0129 requires to introduce the additional work described in theAirbus SB A380-36-8026 Rev2, as Rev1 (already mandated through AD2018-0100) didn't achieve the full intent of the desired modification. The Technical Adaptations ref 80445344/014/2018, 80445344/068/2019<(>,<)> 80599250/019/2019, 8			2020-0129	11/09/2020	Active
10074434		AIRBUS S.A.S.	A321-231/-232 A321-131/-211/-212/-213 A320-214/-231/-232/-233 A319-111/-112/-115/-132/- 133 A318-112/-122		EASA AD 2020-0103 was issued to address a lack of protection againstcurrent injection of 28 Volts DC or 115 Volts AC ona non-rechargeableELT lithium battery, by the introduction of a modification in accordancewith the Airbus Service Bulletins instructions.Since the issuance of the EASA AD 2020-0103		- Airbus Service Bulletin A320-25-1BQN original issue, dated 05 December2019 Airbus Technical Adaptation 80724343/009/2020 issue 1, dated 20 May2020	2020-0103	30/09/2020	Active





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10075670		AIRBUS S.A.S.	A321-271N/-271NX/-272N /-272NX A321-253N/-251NX/-252N X/-253NX A321-213/-231/-232/-251N /-252N A321-111/-112/-131/-211/- 212 A320-253N/-271N/-272N/- 273N A320-253N/-271N/-272N/- 273N A320-231/-232/-233/-251N /-252N A320-211/-212/-214/-215/- 216 A319-131/-132/-133/-151N /-153N A319-111/-112/-113/-114/- 115 A318-111/-112/-121/-122	All manufacturer serial numbers.	The EASA AD 2020-0274 has been issued to reduce the risk ofbatteriesover- discharged and to keep them in a serviceable condition despite longaircraft parking situation.Mandatory Airbus AOT A24N006-20 provides operators with instructions and recommendations for battery preservation, through Off-Wing o		Airbus Repair and Design Approval Form, ref : 0828936/042/2021 Issue A,dated 17 February 2021.	2020-0274	18/02/2021	Active
10075831	REV. 1	AIRBUS S.A.S.	A310-304	There is no change in the compliance time required in the AD 2015-0174(i.e. compliance time of 48 months after respectively SBs A310- 22-2063/ A310-22-2064 first issue date (revision 00)).	EASA AD 2015-0174, covering A300-600/A310 aircraft, mandates correctiveactions requiring the installation and activation of the Stop RudderInputs Warning (SRIW).In particular, EASA AD 2015-0174 requires to install the SRIW device, inaccordance with the instructions of2 Airbus SBs A310-22-2063 and A3		Airbus Service Bulletins A310-22-2067 Rev 00 dated 27 Nov. 2020 andA310-22-2068 Rev 00 dated27 Nov. 2020, or later approved revision.	2015-0174	16/03/2021	Active
10075844	REV. 2	AIRBUS S.A.S.	A321-251N/-253N/-271N/- 272N A321-211/-212/-213/-231/- 232 A320-233/-251N/-271N A319-133, A320-214/-216/-232 A319-111/-112/-115/-131/- 132 A318-112/-121/-122	Manufacturer serial numbers as listed in the applicable SBs, as definedin the AD;	AD 2019-0069 require the modification of the attachments oftheleft-hand and right- hand overhead stowage compartments (OHSC), inaccordance with the instructions of the SB25-1BGK on A320 family CEOaircraft or SB25-1BGL on A320 family NEO aircraft fitted with enhancedcabin modification 35812/ K17239 to		Memo E25M20002933 dated May 28, 2020SB25-1BKTSB25-1BK MSB25-1BKS.(The use of original issue or later approved revisions of theabove-mentioned documents is acceptable for compliancewith theparagraph (1) of the EASA AD 20019-0069.)	2019-0069	05/04/2023	Active
10075996		AIRBUS S.A.S.	ALL		AIRBUS Design Organisation has introduced on February 2021 a Repair andDesign Approval Form (RDAF) document which merges and supersedes theformer Repair Design Approval Sheet (RDAS) and Technical Adaptation (TA) documents.As RDAS and TA acronyms were used in some past EASA AirworthinessDirectives, in			LISTED IN TABLE 1 & 2	30/03/2021	Active
10076331		AIRBUS S.A.S.	A380-861 A380-842 A380-841	The aircraft whose pylons were modified in accordance with theinstructions detailed in RDAFs number80885156/005/202 1, 80885156/007/2021, 80885156/008/2021,80885 156/009/2021, 80885156/011/2021,80885 156/012/2021, 80885156/013/2021, 80894228/004/2021,80894 228/005/2021, 80894228/006/2	ATA 36 – Pneumatic – Pylon / Wing Interface Bleed Duct and Fuel Pipe – InspectionAD 2020-0286 requires to inspect the pylon bleed duct installationwithin 50 FC / 400 FH (whatever occurs first) after 04 August 2020(effective date of EASA AD 2020-0162<(>,<)> superseded by EASA AD2020-0286), and thereaf		None	2020-0286	27/04/2021	Active





EASA Certificate	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
Number 10076777		AIRBUS S.A.S.	A321-271N/-271NX/-272N A321-251N/-251NX/-253N /-253NX A320-251N/271N A319-153N		EASA AD 2020-0103 was issued to address a lack of protection againstcurrent injection of 28 Volts DC or 115 Volts AC ona non-rechargeableELT lithium battery, by the introduction of a modification in accordancewith the Airbus Service Bulletins instructions.Since the issuance of the EASA AD 2020-0103		- Airbus Service Bulletin A320-25-1BQN original issue, dated 05 December2019 Airbus Service Bulletin A320-25-1BQP original issue, dated 05 December2019 Airbus Technical Adaptation 80832689/007/2020 issue 2, dated 29October 2020.	2020-0103	29/06/2021	Active
10077198		AIRBUS S.A.S.	A319-115	None	The AD 2016-0212 was published on 25 October 2016 to address thepotential risk associated to the penetration through the cabin floor ofa vertical strut at Frame (FR) 65, followingan airframe ground contactabove certified vertical speed/ vertical acceleration.To address this potential unsafe conditio		Alternative Method of Compliance 300022876/60079830Airbu smodification 153724 implemented in the Service Bulletins: AirbusSB A320-53-1262 Revision 01 dated 29 July 2016. Airbus SB A320-53-1333original issue dated 29 July 2016 and/or Airbus SB A320-53-1334 originalissue dated 29 July 2016.STC 10054	AD2016-0212	08/09/2021	Active
10077388		AIRBUS S.A.S.	A321-272N/-272NX A321-253N/-253NX/-271N /-271NX A321-251N/-251NX/-252N /-252NX A320-272N/-273N A320-251N/-252N/-253N/- 271N A319-151N/-153N/-171N		The Airworthiness Directive 2021-150C1 mandates the amendment of theAirbus A319/A320/A321 Aircraft Flight Manual by incorporating the AFMTemporary Revision, ref : TR 787 Issue 1,together with the update ofthe Master Minimum Equipment List (MMEL) with the MMEL item 34-23-02B,ident MI-34-23-00008619		Modification 169275MMEL item 34-23-02A, ident MI-34-23-0008619.00050 01A319/A320/A321 AFM TR 787 Issue 1MMEL item 34-23-02B, ident MI-34-23-00008619.00090 01.	AD 2021-0150	01/10/2021	Active
10077415	REV. 1	AIRBUS S.A.S.	A321-272N/-272NX A321-253N/-253NX/-271N /-271NX A321-251N/-251NX/-252N /-252NX A320-272N/-273N A320-251N/-252N/-253N/- 271N A319-151N/-153N A319-111/-112/-113/-114/- 115	This AMOC approval is re- issued at revision 1 to correct sometypographical errors from original issue. The AMOC issue1 supersedes theAMOC original issue.	The Airworthiness Directive 2020-0118 mandates the amendment of theAirbus A319/A320/A321 Aircraft Flight Manual by incorporating the AFMTemporary Revisions, ref : TR 784 Issue 1 and TR 785 Issue 1(respectively applicable to A319NEO Leap1A,and A319CEO CFM56 / A320NEO/A321NEO).Since this AD issuance,		Modification 166790Modification 163903Modification 168506Modification 168507 Issue 2A319/A320/A321 AFM TR 785 Issue 1A319 AFM TR 784 Issue 1	AD 2020-0118	19/10/2021	Active
10077686		AIRBUS S.A.S.	A321-271N/-271NX/-272N /-272NX A321-251NX/-252NX/-253 N/-253NX A321-213/-231/-232/-251N /-252N A321-111/-112/-131/-211/- 212 A320-253N/-271N/-272N/- 273N A320-231/-232/-233/-251N /-252N A320-211/-212/-214/215/- 216 A319-131/-132/-133/-151N /-153N A319-111/-112/-113/-114/- 115 A318-111/-112/-121/-122	Equipment / Furnishings – Emergency Locator Transmitter Protection – Modifications	The EASA AD 2020-0274 has been issued to reduce the risk ofbatteriesover- discharged and to keep them in a serviceable condition despite longaircraft parking situation.Mandatory Airbus AOT A24N006-20 provides operators with instructions and recommendations for battery preservation, through Off-Wing o		Airbus Repair and Design Approval Form, ref : 80931795/002/2021 Issue A,dated 12 July 2021.	AD 2020-0274	12/11/2021	Active





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0078244		AIRBUS S.A.S.	A380-842	- MSNs 015 and 026 ORSs are to be inspected in accordance withrequirement (1) of AD 2022-0019, with the possibility of taking creditof what allowed under the provisions of (3) of the same AD 2022-0019 In analogy to what required by (1) of AD 2022-0019, OIFSs and OFSsinspections carried out in acco	EASA AD 2022-0019 requires accomplishing a special detailedinspection(SDI) of the LH/RH wing outer rear spar (ORS), outer inner front spar(OIFS) and outer front spar (OFS) at theindicated locations, beforeexceeding 180 months since wing box assembly and thereafter at intervalsnot to exceed 36 mont		RDAF 80946955/007/2021RDAF 80979461/008/2021.	2022-0019	01/02/2022	Active
0079379		AIRBUS S.A.S.	A320-271N/-272N/-273N A320-251N/-252N/-253N	All manufacturer serial numbers, equipped with mod 162443.	The EASA AD 2020-0125 was retaining the requirements of theEASA AD2019-0189, implementing Temporary Revision of AFM tolimit the Center ofGravity envelop of the affected aircraftdue to a reduced efficiency of the angle of attack protection when the aeroplane is set in certainflight configurations a		EASA MAJOR CHANGE APPROVAL 10078160.	EASA 2020-0125	31/05/2022	Active
0079527		AIRBUS S.A.S.	A321-271NX/-271NX A321-252NX/-253NX/-271 N/-272N A321-251N/-252N/-253N/- 251NX	All manufacturer serial numbers, equipped with mod 162443.	Alternative Method of Compliance to Airworthiness DirectiveEASA2019-0171R2.The EASA AD 2019-0171R2 was retaining the requirements of the EASA AD2019-0171 (original issue and Revision 1), implementing TemporaryRevision of AFM to limit the Center of Gravity envelop of the affectedaircraft due to a re		EASA MAJOR CHANGE APPROVAL 10078160.	EASA 2019-0171R2	20/06/2022	Active
0079577		AIRBUS S.A.S.	A321-271NX/-272NX A321-271N/-272N A321-251NX/-252NX/-253 NX A321-251N/-252N/-253N A321-213/-231/-232 A321-111/-112/-131/-211/- 212 A320-271N/-272N/-273N A320-251N/-252N/-253N A320-251N/-252N/-253N A320-231/-232/-233 A320-211/-212/-214/-215/- 216 A319-151N/-153N/-171N A319-151N/-153N/-171N A319-131/-132/-133 A319-111/-112/-113/-114/- 115 A318-111/-112/-121/-122		Time Limits / Maintenance Checks – Airworthiness Limitations Section –Part 2 – Damage Tolerant Airworthiness LimitationItems – ImplementationSee EASA Approved Annex		Airbus SB A320-53-1383.Airbus SB A320-53-1372.Airbus SB A320-53-1268.(any later approved revison of those technical documents are acceptable)	2022-0085	28/06/2022	Active
0080483		AIRBUS S.A.S.	A321-272N/-272NX A321-271N/-271NX A321-253N/-253NX A321-252N/-252NX A321-251N/-251NX A321-251N/-251NX A321-231/-232 A321-211/-212/-213 A321-111/-112/-131 A320-271N/-272N/-273N A320-251N/-252N/-253N A320-251N/-252N/-253N A320-231/-232/-233 A320-211/-212/-214/-215/- 216 A319-151N/-153N/-171N A319-131/-132/-133 A319-111/-112/-113/-114/- 115 A318-111/-112/-121/-122	None	ATA 25 – Equipment / Furnishings – Galleys – InspectionThe AD 2021-0183R1 mandates an inspection regime on specific galleys,and, depending upon findings, the accomplishment of corrective actionsSince the effective date of this AD, Airbus has issued, on August 12,2022< (>,<)> the Service Bulletins 25		None	2021-0183R1	31/10/2022	Active

This report lists certificates issued during the period 01/01/2020 until 22/04/2025. In case you need a certificate not included, please contact applicant.services@easa.europa.eu. Page 21 of 68





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10080566	REV. 2	AIRBUS S.A.S.	A321-271N/-272N/-271NX /-272NX A321-251NX/-252NX/-253 NX A321-251N/-252N/-253N A321-211/-212/-213/-231/- 232 A321-111/-112/-131 A320-271N/-272N/-273N A320-251N/-252N/-253N A320-251N/-252N/-253N A320-231/-232/-233 A320-211/-212/-214/-215/- 216 A319-151N/-153N/-171N A319-131/-132/-133 A319-111/-112/-113/-114/- 115 A318-111/-112/-121/-122	None	ATA 32 – Landing Gear – Braking and Steering Control Unit –Replacement/ Master Minimum Equipment List – AmendmentThe AD2022-0032R1 mandates the accomplishment of the AOT A32N025-22which prescribes not to install the BSCU P/N E21327307 (BSCU affectedpart) onto aircraft not having been delivered with		Airbus Service Bulletin (SB) A320-92-1149Airbus Modification 171984Airbus AOT A32N025-22	2022-0032R1	16/01/2023	Active
10081616		AIRBUS S.A.S.	A321-271N A321-213/-231/-232/-251N /-253N A321-111/-112/-131/-211/- 212 A320-231/-232/-233/-251N /-271N A320-211/-212/-214/-215/- 216 A319-131/-132/-133 A319-111/-112/-113/-114/- 115 A318-111/-112/-121/-122	Aeroplane previously modified in accordance with Airbus SB A320-34-1760at original issue and for which the compliance to the AD 2017-0091R2paragraph (1) has been achieved through the AMOC approval 10073967 donot require any further action.	EASA AD 2017-0091R2 was issued to address the number of false TCASresolution advisories (RA) by modifying the traffic collision avoidancesystem (TCAS) processor through an upgradeof the software, or thereplacement of the Honeywell TCAS TPA-100B processor having the P/ N940-0351-001 with a TCAS TPA-1		- Airbus SB A320-34-1760 dated September 09, 2019, for A320CEO models- Airbus SB A320-34-1761 dated September 09, 2019, for A320 NEO models Airbus modification 166196The use oflater approved revisions of these documents is acceptable.	2017-0091R2	30/03/2023	Active
10081734		AIRBUS S.A.S.	A300 B4-622R	Limited to A300 B4-622R aircraft converted from Pax to Freighter versionthrough Airbus STC EASA.A.S.00387 (10013945) atany revision.	As a result of the Extended Service Goal 2 exercise (ESG2) it was shownthat the door frame shells of passenger doors 2 and 4 (both sides of theaeroplane) may not have sufficient structural length to enable theaeroplane to operate safety beyond ESG1 (42 500 FC or 89 000 FH) and upto ESG2 (51 000 FC o		Airbus Service Bulletin SB A300-53-6170 Rev. 01 dated 17 February 2023Airbus Engineering Order PFM06100A	2012-0044	19/04/2023	Active
10083084		AIRBUS S.A.S.	A321-272NX A321-272N A321-271NX A321-271N A320-273N A320-272N A320-271N A319-171N	None	ATA 29 – Hydraulic Power – Pylon / Engine Interface Rods – Inspection / ReplacementThe Airbus modification 171858 introduces a new rods design (EASA MajorChange Approval 10082722, dated 4 September 2023).For rods post modification 171858, there is no more need to inspect asper the AD 2021-0177R1, bec		None	2021-0177R1	24/10/2023	Active
10083571		AIRBUS S.A.S.	A380-861	This AMOC is valid for a period of three months after the date of suance.	Occurrences were reported of finding cracks in wing spars of several inservice A380 aeroplanes.To restore the structural integrity of the wing, EASA issued AD2023-0198 requesting periodical inspection of the spars.MSN 140 & 182 wing outer rear spars will be shortly inspected by PAUT(Phased Array Ult		Application ref. L57D23035514 issue 1 dated 15/12/2023.	2023-0198	19/12/2023	Active





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10083785		AIRBUS S.A.S.	A300F4-608ST		Mandatory Airbus A300-600 SBs A300-53-6011 and A300-53-6022 requestinitial and repetitive inspections of areas behind scuff plates atpassenger/crew doors and bulk cargo door for corrosion and crackdetection.They were made mandatory for theBeluga ST (A300F4-608ST) model througha dedicated A300-600ST		Airbus Service Bulletin SB A300-53-6011 Rev. 08 dated 28 April 2008Airbus Service Bulletin SB A300-53-6022 Rev. 05 dated 28 April 2008The use of later approved revisions of the above-mentioned documents isacceptable for compliance with therequirements of this AMOC.	2000-316-044(B)	25/01/2024	Active
10084322		AIRBUS S.A.S.	A321-271NX/-272NX A321-251NX/-252NX/-253 NX		EASA issued the AD 2022-0176 mandating the removal of the protectivecover fitted on the orifice of the slide reservoir regulator valve.In the meantime, Airbus has introduced threeproduction design changesthat introduce a modified orifice to increase the availability of theventing function.EASA cons		Design change 173621: E/ F – ESCAPE FACILITIES - CABIN - INSTALL MODIFIEDPRESSURE RESERVOIR IN CARGO COMPARTMENT FOR OWS (PN70200-101-102modified to 70200-109 -110).Design change 173622: E/F – ESCAPE FACILITIES - CABIN - INSTALL MODIFIEDPRESSURE RESERVOIR FOR OHSC VARIANT (PN 70200-103 - 104 modified	EASA 2022-0176	22/04/2024	Active
10084925		AIRBUS S.A.S.	A300F4-608ST	This approval shall remain valid as long as the referenced AD is notcancelled or superseded.This approval shall remain valid as long as the referenced MRD CPCP taskis not cancelled or revised in a way that the inspection requirementwould be relaxed (eg. smaller inspected area or different inspection	Several cases of corrosion on the lower wing root joint, located in thewing bottom skin inboard and outboard of the external lower surfacesplice, have been reported by operators. This condition, if not detected and corrected, could affect the structural integrity of the airframe.Airbus Service Bullet		- Technical Note K5710RP2400384 Issue 1.1 dated 04-APR-24 A300-600ST -Analysis of applicability of A300-600 ISB 57-6047on Beluga fleet- A300F4-608ST Maintenance Requirements Document (MRD) (Rev. 4) CPCPtask 571066-01-1.	98-041-013(B)	16/07/2024	Active
10086556		AIRBUS S.A.S.	A321-271N/-271NX/-272N /-272NX A321-253N/-253NX/-253N Y A321-251N/-251NX/-252N /-252NX A320-272N/-273N A320-251N/-252N/-253N/- 271N A319-151N/-153N/-171N/- 173N		AD 2025-0037 requires an amendment to the AFM by incorporating theprocedures as described in the OEB 63 issue 1.0 (dated 07 February2025). This amendment includes instructions forregaining communicationand transponder functionality under certain failure conditions, foraeroplanes fitted with Digital			2025-0037	27/02/2025	Active
10086755		AIRBUS S.A.S.	A380-841	None	ATA 57 – Wings – Wing Front and Rear Spars – InspectionOccurrences were reported of finding cracks in wing spars of several inservice A380 aeroplanes.To restore the structural integrity of the wing, EASA issued AD2024-0191 requesting periodical inspection of the spars.MSN 00085 wing outer front andi		Application ref. L57D25009994 issue 1 dated 20/03/2025	2024-0191	24/03/2025	Active



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10079831		ALITALIA SOCIETA AEREA ITALIANA S.p	777F 777-300ER 777-300 777-200LR 777-200	None	Landing Gear - Main Landing Gear (MLG) Inner Cylinder / Truck Beam -Inspection / LubricationAlitalia applied to EASA for an AMOC to AD 2012-19-10.EASA considers that Alitalia's proposal to grant a new one-timeexemption interval of 4125 days, which allows to phase the requiredinspection with the next		Application ref. 2019-001/ VLS/AMOC issue 1 dated 05/06/2022	2012-19-10	28/07/2022	Active
0074379	REV. 1	APTOZ EHF.	737-400	This AMOC only applies to serial numbers 24436.This AMOC isvalid until 23 July 2021.	This AMOC provides an acceptable level of safety to replacethe actionsidentified in the AD 2012-23-04 paragraph (p) for FCBS STA 727 Bulkhead- Webs, chords and stiffeners (53-021).AD 2012-23-04 was issued to detect and correct fatigue cracking of thebody station (BS) 727 frame outboard chord, and c		V1 Aerospace Repair Instructions R20-00774-001 Rev 03.FAA AMOC approval 790-20-10058.	2012-23-04	29/09/2020	Active
10074380	REV. 1	APTOZ EHF.	737-400	This AMOC only applies to serial numbers 24436.This AMOC isvalid until 23 July 2021.	This AMOC provides an acceptable level of safety to replacethe actionsidentified in the AD 2018-01-08.AD 2018-01-08 requires repetitive high frequency eddy current (HFEC) anddetailed inspections to detect cracks in the auxiliary chord radius;time-limited repetitive inspections; replacement of the o		V1 Aerospace Repair Instructions R20-00774-001 Rev 03.FAA AMOC approval 790-20-10058.	2018-01-08	29/09/2020	Active
0074381	REV. 1	APTOZ EHF.	737-400	This AMOC only applies to serial number 24436.This AMOC is valid until 23 July 2021	This AMOC provides an acceptable level of safety to replacethe actionsidentified in the AD 2016-11-04 paragraph (o) for FCBS Fuselage LowerLobe Frames BS 188-520 (737-CLB-FCBS 53-010).AD 2016-11-04 requires inspections, modification and corrective actionsfor cracking at a wirepenetration hole.The a		V1 Aerospace Repair Instructions R20-00776-001, Rev 02.FAA AMOC approval 790-20-10059.	2016-11-04	29/09/2020	Active
10074400	REV. 1	APTOZ EHF.	737-400	This AMOC only applies to serial number 24436.This AMOC is valid until 23 July 2021.	This AMOC provides an acceptable level of safety to replace the actions identified in the AD 2018-19-20 paragraph (m)(1) for FCBS stub beam atLH side body station (STA) 616 .AD 2018-19-20 requires inspections and corrective actions for fatiguecracking of certain fuselage frames and stub beams and pos		V1 Aerospace Repair Instructions R20-00778-001, Rev 06.FAA AMOC approval 790-20-10072.	2018-19-20	29/09/2020	Active
10074996		APTOZ EHF.	737-400	Accomplish inspections of the repair area in accordance with theinstructions of and within the thresholds and intervalsas identified indocument D20-00454-001 Rev 01. All other areas must be inspected asmandated by FAA AD 2017-20-14.All provisions of AD 2017-20-14 that are not specifically reference	Doors, Fuselage, Nacelles/Pylons, Stabilizers, Wings - StructuralSignificant Items - Inspections / Maintenance or Inspection ProgramRevisionAD 2017-20-14 was issued to provide supplemental inspections that arerequired for timely detection of fatigue cracking for certain structuralsignificant items (V1 Aerospace LLC D20-00454-001 Rev 01FAA AMOC approval 790-20-11089a	(FAA) AD 2017-20-14	27/11/2020	Active





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10074997		APTOZ EHF.	737-400	Accomplish inspections of the repair area in accordance with theinstructions of and within the thresholds and intervalsas identified indocument D20-00454-001 Rev 01. All other areas must be inspected asmandated by FAA AD 2016-20-02.All provisions of AD 2016-20-02 that are not specifically reference	AD 2016-20-02 was issued to detect and correct fatigue cracking of theaft pressure bulkhead web at the "Y"-chord, which could result inreduced structural integrity of the airplane and rapid decompression ofthe fuselage.During a C-check (25 May 2011), corrosion damage has been detected onthe aft pr		V1 Aerospace LLC D20-00454-001 Rev 01FAA AMOC approval 790-20-11089a	(FAA) AD 2016-20-02	27/11/2020	Active
10074998		APTOZ EHF.	737-400	Accomplish inspections of the repair area in accordance with theinstructions of and within the thresholds and intervalsas identified indocument D20-00454-001 Rev 01. All other areas must be inspected asmandated by FAA AD 2012-18-13R1.All provisions of AD 2012-18-13R1 that are not specifically refer	AD 2012-18-13R1 was issued to detect and correct fatigue cracks in theaft pressure bulkhead, which could result in rapid decompression of thefuselage.During a C-check (25 May 2011), corrosion damage has been detected onthe aft pressure bulkhead (APB) web, the APB y-chord frame and the APBy-chord ang		V1 Aerospace LLC D20-00454-001 Rev 01FAA AMOC approval 790-20-11089a	(FAA) AD 2012-18-13R1	27/11/2020	Active
10075133	REV. 1	APTOZ EHF.	737-400	- All provisions of AD 2016-11-04 and AD 2017-2014 that arenotspecifically referenced above remain fully applicable and must becomplied with accordingly This approval is applicable to Boeing 737-400 aircraft, SN 24436.	2016-11-04 requires inspections, modification and corrective actions forcracking at a wire penetration hole. AD 2017-20-14 requires timelydetection of fatigue cracking for certainstructural significant items.During a scheduled accomplishment of Boeing Alert Service Bulletin (SB)737-53A1279, Rev 02,		- V1 Aerospace LLC Damage Tolerance analysis D20-00776-001 Rev 00, 12August 2020- FAA AMOC approval letter 790-20-12458, dated 22 September 2020	AD 2016-11-04& AD 2017-20-14	16/12/2020	Active
10075541		APTOZ EHF.	737-400	Accomplish the actions identified in, and within the threshold asspecified in, section 5.0 – Instructions for ContinuedAirworthiness – of document D20-00774-001 Rev 00.All provisions of AD 2018-01-08 that are not specifically referencedabove remain fully applicable and must be complied with accordin	Fuselage - Frame Outboard Chord / Auxiliary Chord - InspectionAD 2018-01-08 requires performing HFEC inspection of the radius of theoutboard chord and the auxiliary chord at STA 727, S-18A in accordancewith Boeing SB 737-53A1166.During inspection per Service Bulletin SB 737-53A1166 (Group 7 airplane		V1 Aerospace LLC D20-00774-001 Rev 00FAA AMOC approval 790-20-15085	(FAA) 2018-01-08	01/02/2021	Active



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10075542		APTOZ EHF.	737-400	Accomplish the actions identified in, and within the threshold asspecified in, section 5.0 – Instructions for ContinuedAirworthiness – of document D20-00774-001 Rev 00.All provisions of AD 2012-23-04 that are not specifically referencedabove remain fully applicable and must be complied with accordin	Fuselage - Frame Body Station (BS) 727 and Stringer 18A Outboard Chords- Inspection / RepairAD 2012-23-04 was issued todetect and correct fatigue cracking of thebody station (BS)727frame outboard chord, and cracks in the radius of the auxiliary chord, which could result inreduced structural integri		V1 Aerospace LLC D20-00774-001 Rev 00FAA AMOC approval 790-20-15085	(FAA) 2012-23-04	01/02/2021	Active
10075734		APTOZ EHF.	737-400	All provisions of FAA AD 2015-21-06 and AD 2017-22-04 that are notspecifically referenced above remain fully applicableand must becomplied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10015732.	AD 2015-21-06 requires inspection of the lower skin at the lower row offasteners in the lap joints of the fuselage, repair of any crackingfound, and modification of the fuselage lap joints at certain locationsin accordance with the Boeing Alert Service Bulletin (SB) 737-53A1177, Revision 7.AD 2017-22		 KF Aerospace R24751.1-053-016-001, Rev NC, dated 9 Oct 2020 KF Aerospace R24751.1-053-017-001, Rev NC, dated 9 Oct 2020 V1 Aerospace D20-00850-001, Rev. 01, dated 12 Dec 2020·V1 Aerospace D20-00851-001, Rev. 01, dated 12 Dec 2020· V1 Aerospace S20-00850-001, Rev. 00, dated 21 Oct 2020· V1 Aeros 	(FAA) 2015-21-06 & 2017-22-04	26/02/2021	Active
10077130		APTOZ EHF.	737-400	All provisions of FAA AD 2015-06-07 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10015732.	Forward Entry Doorway and Airstairs Cutout - InspectionAD 2015-06-07 was issued to detect and correct cracks in the lowercorners of the forward entry door skin cutout and the upper corners ofthe airstairs cutout, which could progress and result in an inability tomaintain cabin pressurization. Inspec		-KF Aerospace R24573.1-053-008-001, Rev NC, dated 19 Jan 2021-V1 Aerospace D21-00972-001, Rev 00, dated 26 Apr 2021-V1 Aerospace S21-00972-001, Rev 00, dated 29 Mar 2021-Boeing Special Attention SB 737-53-1083 Rev 5, dated 22 Jul 2014- FAA AMOC Approval Letter 790-21-7458, dated 26 May 2021	(FAA) 2015-06-07	24/08/2021	Active
10077132		APTOZ EHF.	737-400	All provisions of FAA AD 2017-22-04 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10015732.	Fuselage - Skin Panels - InspectionAD 2017-22-04 was issuedto detect and correct disbonded skin panels,which could result in fuselage skin cracking, rapid decompression, andloss of structural integrity of the airplane. Inspection to detect andcorrect disbonded skin panels are in accordance with Boe		-KF Aerospace R24573.1-053-008-001, Rev NC, dated 19 Jan 2021-V1 Aerospace D21-00972-001, Rev 00, dated 26 Apr 2021-Boeing Alert SB 737-53A1349 Rev IR, dated 23 Aug 2016- Boeing Special Attention SB 737-53-1083 Rev 5, dated 22 Jul 2014-FAA AMOC Approval Letter 790-21-7458, dated 26 May 2021	(FAA) 2017-22-04	24/08/2021	Active



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10077133		APTOZ EHF.	737-400	All provisions of FAA AD 2017-20-14 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10015732.	Doors, Fuselage, Nacelles/Pylons, Stabilizers, Wings - StructuralSignificant Items - Inspections / Maintenance or Inspection ProgramRevisionAD 2017-20-14 mandates that supplemental inspections are required fortimely detection of fatigue cracking for certain structural significantitems (SSIs) in acco		-KF Aerospace R24573.1-053-008-001, Rev NC, dated 19 Jan 2021-V1 Aerospace D21-00972-001, Rev 00, dated 26 Apr 2021-Boeing Supplemental Structural Inspection Document (SSID), Models737-300/400/500 Airplanes, Report D6-82669, Rev Oct 2015- Boeing Special Attention SB 737-53-1083 Rev 5, dated 22 Jul 20	(FAA) 2017-20-14	24/08/2021	Active
10077689		APTOZ EHF.	737-400	- All provisions of FAA AD 2016-11-04 that are not specificallyreferenced above remain fully applicable and must be complied withaccordingly Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10016227.	AD 2016-11-04 was issued to detect and correct fatigue cracking of thefuselage frames and frame reinforcements, which could result in reducedstructural integrity of the airplane. The FAA, through FAA AMOC Approval Letter 790-21-15349, has alreadyapproved to use V1 Aerospace Repair Instructions R21-01		- V1 Aerospace Repair Instructions R21-01048-001, Rev 05, dated 16 Jun2021- V1 Aerospace Static Substantiation S21-01048-001, Rev 00, dated 17 Jun2021- V1 Aerospace Damage Tolerance Substantiation D21-01048-001, Rev 00,dated 13 Oct 2021- FAA AMOC Approval Letter 790-21-8932, dated 23 Jun 2021- FAA A	AD 2016-11-04	15/11/2021	Active
10077692		APTOZ EHF.	737-400	All provisions of FAA AD 2017-20-14 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10016227.	Doors, Fuselage, Nacelles/Pylons, Stabilizers, Wings - StructuralSignificant Items - Inspections / Maintenance or Inspection ProgramRevisionAD 2017-20-14 mandates that supplemental inspections are required fortimely detection of fatigue cracking for certain structural significantitems (SSIs) in acco		-V1 Aerospace Repair Instructions R21-01048-001, Rev 05, dated 16 Jun2021-V1 Aerospace Damage Tolerance Substantiation D21-01048-001, Rev 00,dated 13 Oct 2021- Boeing Supplemental Structural Inspection Document (SSID), Models737-300/400/500Air planes, Report D6-82669, Revision October 2015- FAA AMOC A	(FAA) 2017-20-14	15/11/2021	Active
10077742		APTOZ EHF.	737-400	- All provisions of FAA AD 2016-11-04 that are not specificallyreferenced above remain fully applicable and must be complied withaccordingly Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10016227.	AD 2016-11-04 was issued to detect and correct fatigue cracking of thefuselage frames and frame reinforcements, which could result in reducedstructural integrity of the airplane.The FAA, through FAA AMOC Approval Letter 790-21-8925, has alreadyapproved to use V1 Aerospace Repair Instructions R21-010		- V1 Aerospace Repair Instructions R21-01047-001, Rev 03, dated 9 Jun2021- V1 Aerospace Static Substantiation S21-01047-001, Rev 00, dated 17 Jun2021- FAA AMOC Approval Letter 790-21-8925, dated 23 Jun 2021.	2016-11-04	22/11/2021	Active





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10077809		APTOZ EHF.	737-400	- All provisions of FAA AD 2017-24-05 that are not specificallyreferenced above remain fully applicable and must be complied withaccordingly Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10015732.	AD 2017-24-05 was issued to detect and correct cracks in the upper aftskin of the wings, which could result in the inability of a principalstructural element to sustain limit load, and consequent reducedstructural integrity of the airplane.During a heavy maintenance visit the operator found corrosio		- V1 Aerospace Repair Instructions R21-01040-001, Rev 00, dated 21 May2021- V1 Aerospace Static Substantiation S21-01040-001, Rev 00, dated 7 Jun2021- V1 Aerospace Damage Tolerance Substantiation D21-01040-001, Rev 00,dated 8 Sep 2021- FAAAMOC Approval Letter 790-21-13943, dated 4 Oct 2021.	(FAA) 2017-24-05	30/11/2021	Active
10077810		APTOZ EHF.	737-400	- All provisions of FAA AD 2019-23-16 that are not specificallyreferenced above remain fully applicable and must be complied withaccordingly Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10015732.	AD 2019-23-16 was issued to address cracks in the upper wing skin, whichcould grow undetected. This condition, if not addressed, could result inthe inability of the structure to carry limit load and adversely affectthe structural integrityof the airplane.During a heavy maintenance visit the operato		- V1 Aerospace Repair Instructions R21-01046-001, Rev 01, dated 3 Jun2021- V1 Aerospace Static Substantiation S21-01046-001, Rev 00, dated 8 Jun2021- V1 Aerospace Damage ToleranceSubstantiation D21-01046-001, Rev 00,dated 8 Sep 2021- FAA AMOC Approval Letter 790-21-13943, dated 6 Oct 2021.	(FAA) 2019-23-16	30/11/2021	Active
10077823		APTOZ EHF.	737-400	All provisions of FAA AD 2017-20-14 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10016227.	Doors, Fuselage, Nacelles/Pylons, Stabilizers, Wings - StructuralSignificant Items - Inspections / Maintenance or Inspection ProgramRevisionAD 2017-20-14 was issued to provide supplemental inspections that arerequired for timely detection of fatigue cracking for certain structuralsignificant items (-V1 Aerospace Repair Instructions R21-01047-001, Rev 03, dated 9 Jun2021-V1 Aerospace Damage Tolerance Substantiation D21-01047-001, Rev 00,dated 6 Oct 2021-FAA AMOC Approval Letter 790-21-14927, dated 22 Oct 2021	(FAA) 2017-20-14	02/12/2021	Active
10077824		APTOZ EHF.	737-400	All provisions of FAA AD 2016-11-04 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10016227.	Fuselage - Frame and Frame Reinforcement - InspectionAD 2016-11-04 was issued to detect and correct fatigue cracking ofthefuselage frames and frame reinforcements, which could result in reducedstructural integrity of the airplane. The FAA, through FAA AMOC Approval Letter 790-21-14927, has alreadyap		-V1 Aerospace Repair Instructions R21-01047-001, Rev 03, dated 9 Jun2021-V1 Aerospace Static Substantiation S21-01047-001, Rev 00, dated 17 Jun2021-V1 Aerospace Damage Tolerance Substantiation D21-01047-001, Rev 00,dated 6 Oct 2021-FAA AMOC Approval Letter 790-21-8925, dated 23 Jun 2021-FAA AMOC App	(FAA) 2016-11-04	02/12/2021	Active



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10077957		APTOZ EHF.	737-400	All provisions of FAA AD 2017-22-04 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10015732.	AD 2017-22-04 requires inspections and corrective actions for disbondedskin doublers on 737 airplanes in accordance with the Boeing Alert (SB)737-53A1349, Rev 00.The FAA, through FAA AMOC Approval Letter 790-21-11041a, has alreadyapproved to use KF Aerospace R24573.1-053-009-001, Rev NC, supported b		- KF Aerospace R24573.1-053-009-001, Rev. NC, dated 8 Apr 2021- KF Aerospace R24573.1-053-006-001, Rev. NC, dated 11 Mar 2021- KF Aerospace R24573.1-053-004-001, Rev NC, dated 12 Apr 2021- V1 Aerospace Damage Tolerance Substantiation D21-00952-001, Rev 00,dated 28 Jun 2021- V1 Aerospace Damage Toler	(FAA) 2017-22-04	15/12/2021	Active
10077977		APTOZ EHF.	737-400	All provisions of FAA AD 2017-20-14 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10015732.	AD 2017-20-14 was issued to provide supplemental inspections that arerequired for timely detection of fatigue cracking for certain structuralsignificant items (SSIs) and to ensure the continued structural integrity of the affected airplanes. Paragraph (h) requires initial and repetitive inspections to		- KF Aerospace R24573.1-053-009-001, Rev. NC, dated 8 Apr 2021- KF Aerospace R24573.1-053-004-001, Rev NC, dated 12 Apr2021 KF Aerospace R737888-052-002-001 Rev NC, dated 26 Apr 2021- V1 Aerospace Damage Tolerance Substantiation D21-00952-001, Rev 00,dated 28 Jun 2021- V1 Aerospace Damage Toleran	(FAA) 2017-20-14	15/12/2021	Active
10077978		APTOZ EHF.	737-400	All provisions of FAA AD 2015-21-06 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10015732.	AD 2015-21-06 requires inspection of the lower skin at the lower row offasteners in the lap joints of the fuselage, repair of any crackingfound, and modification of the fuselage lap joints at certain locationsin accordance with Boeing Alert Service Bulletin (SB) 737-53A1177, Rev7.During maintenance1		- KF Aerospace R24573.1-053-009-001, Rev. NC, dated 8 Apr 2021- V1 Aerospace Damage Tolerance Substantiation D21-00952-001, Rev 00,dated 28 Jun 2021- Boeing Alert Service Bulletin(SB) 737-53A1177, Rev 7, dated 14 Jun2013- FAA AMOC Approval Letter 790-21-11041a, dated 12 Aug 2021.	(FAA) 2015-21-06	15/12/2021	Active
10080634	REV. 1	APTOZ EHF.	737-800	- All provisions of AD 2013-19-23 that are not specificallyreferencedabov e remain fully applicable and must be complied with accordingly Only applicable to Boeing 737-800 airplane, s/n 28642, modified tofreighter in accordance with EASASTC 10076409 Additional limitations as per V1 Aerospace do	AD 2013-19-23 requires the implementation of additional structuralinspection of PSEs for affected aircraft. On a/c 737-800, s/n 28642, adamage has been detected in April 2016, between STA 787 and 807, STR 18Rand 20R, and repaired iaw Boeing approved data. That repair required afurther assessment toA		V1 Aerospace document D22-01526-001 Rev. 01FAA AMOC approval letter 782-22-11172, dated 20 October 2022.	(FAA) 2013-19-23	25/04/2023	Active





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10081846		APTOZ EHF.	737-400	- All provisions of AD 2017-20-14 that are not specificallyreferencedabov e remain fully applicable and must be complied with accordingly Only applicable to Boeing 737-400 airplane, s/n 25110, modified tofreighter in accordance withEASA STC 10015732 (FAA STC ST01827LA) Additional limitations as0	AD 2017-20-14 mandates that supplemental inspections are required fortimely detection of fatigue cracking for certain structural significantitems (SSIs) in accordance with Boeing Supplemental StructuralInspection Document (SSID), Models 737-300/400/500 Airplanes, ReportD6-82669, Revision October 201		- V1 Aerospace Repair R21-01117-001, Revision 01- V1 Aerospace Damage Tolerance Substantiation D21-01117-001, Revision00- FAA AMOC approval letter 790-21-17763.	(FAA) AD 2017-20-14	05/05/2023	Active
10081847		APTOZ EHF.	737-400	- All provisions of AD 2017-02-05 that are not specificallyreferencedabov e remain fully applicable and must be complied with accordingly Only applicable to Boeing 737-400 airplane, s/n 25110, modified tofreighter in accordance withEASA STC 10015732 (FAA STC ST01827LA) Additional limitations as0	AD 2017-02-05 requires inspections and repairs for crackingof thefuselage skin assembly and the bear strap at the corners of the aftentry and aft galley doorways.On 737-400 s/n 25110, during heavy maintenance inspection crackingdamage wasidentified at the fwd lower corner of the LH aft entry doorc		- V1 Aerospace Repair R21-01117-001, Revision 01- V1 Aerospace Damage Tolerance Substantiation D21-01117-001, Revision00- FAA AMOC approval letter 790-21-17763.	(FAA) AD 2017-02-05	05/05/2023	Active
10081848		APTOZ EHF.	737-400	- All provisions of AD 2010-05-13 that are not specificallyreferencedabov e remain fully applicable and must be complied with accordingly Only applicable to Boeing 737-400 airplane, s/n 25110, modified tofreighter in accordance withEASA STC 10015732 (FAA STC ST01827LA) Additional limitations as0	AD 2010-05-13 was issued to prevent rapid decompression of the airplanedue to fatigue cracks resulting from scribe lines on pressurizedfuselage structureOn 737-400 s/n 25110, during heavy maintenance inspection crackingdamage was identified at the fwd lower corner of the LH aft entry doorcutout. The		- V1 Aerospace Repair R21-01117-001, Revision 01- V1 Aerospace Damage Tolerance Substantiation D21-01117-001, Revision00- FAA AMOC approval letter 790-21-17763.	(FAA) AD 2010-05-13	05/05/2023	Active
10081849		APTOZ EHF.	737-400	- All provisions of AD 2003-14-06 that are not specificallyreferencedabov e remain fully applicable and must be complied with accordingly Only applicable to Boeing 737-400 airplane, s/n 25110, modified tofreighter in accordance withEASA STC 10015732 (FAA STC ST01827LA) Additional limitations as0	AD 2003-14-06 requires inspection and corrective actions for cracks atthe fuselage skin at the upper row of fasteners on all lap joints frombody station (BS) 259 to BS 1016 and also provides a terminating actionfor some aircraft.On 737-400s/n 25110, during heavy maintenance inspection crackingdamag		- V1 Aerospace Repair R21-01117-001, Revision 01- V1 Aerospace Damage Tolerance Substantiation D21-01117-001, Revision00- FAA AMOC approval letter 790-21-17763.	(FAA) AD 2003-14-06	05/05/2023	Active
10081850	REV. 1	APTOZ EHF.	737-400	- All provisions of AD 2005-19-25 that are not specificallyreferencedabov e remain fully applicable and must be complied with accordingly Only applicable to Boeing 737-400 airplane, s/n 25111, modified tofreighter in accordance withEASA STC 10015732 (FAA STC ST01827LA) Additional limitations as0	AD 2005-19-25 paragraph (g) requires inspections and corrective actionsfor cracks of the countersunk rivet holes in thelower lobe adjacent tothe radio altimeter cutouts as well as at an adjacent exhaust port ductcutout. AD 2005-19-25 paragraph (I) also provides an option for apreventive modificatio		V1 Aerospace document R21-01144-001 Rev. 01V1 Aerospace Damage Tolerance Substantiation D21-01144-001, Revision 00FAA AMOC approval letter 790-21-15779.	(FAA) AD 2005-19-25	10/05/2023	Active





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0081852		APTOZ EHF.	737-400	- All provisions of AD 2003-14-06 that are not specificallyreferencedabov e remain fully applicable and must be complied with accordingly Only applicable to Boeing 737-400 airplane, s/n 25097, modified tofreighter in accordance withEASA STC 10015732 (FAA STC ST01827LA) Additional limitations as0	AD 2003-14-06 requires inspection and corrective actions for cracks atthe fuselage skin at the upper row of fasteners on all lap joints frombody station (BS) 259 to BS 1016.On 737-400 s/n 25097, an undocumented repair has been found at STA867-877, STR 1-2L, under dorsal fin landing area, not conform		- V1 Aerospace Repair R21-01067-001, Revision 02- V1 Aerospace Damage Tolerance Substantiation D21-01067-001, Revision01- FAA AMOC approval letter 792-22-00129.	(FAA) AD 2003-14-06	05/05/2023	Active
0081853		APTOZ EHF.	737-400	- All provisions of AD 2017-20-14 that are not specificallyreferencedabov e remain fully applicable and must be complied with accordingly Only applicable to Boeing 737-400 airplane, s/n 25097, modified tofreighter in accordance withEASA STC 10015732 (FAA STC ST01827LA) Additional limitations as0	AD 2017-20-14 requires inspections and corrective actions for cracks inSSIs in accordance with the October 2015 revision of the Boeing SSIDD6-82669.On 737-400 s/n 25097, an undocumented repair has been found at STA867-877, STR 1-2L, under dorsal fin landing area, not conforming to anyrepair per Boin		- V1 Aerospace Repair R21-01067-001, Revision 02- V1 Aerospace Damage Tolerance Substantiation D21-01067-001, Revision01- FAA AMOC approval letter 792-22-00129.	(FAA) AD 2017-20-14	05/05/2023	Active
10081854		APTOZ EHF.	737-400	- All provisions of AD 2017-22-04 that are not specificallyreferencedabov e remain fully applicable and must be complied with accordingly Only applicable to Boeing 737-400 airplane, s/n 25097, modified tofreighter in accordance withEASA STC 10015732 (FAA STC ST01827LA) Additional limitations as0	AD 2017-22-04 requires inspections and corrective actions for disbondedskin doublers on 737 airplanes iaw Boeing Alert Service Bulletin (SB) 737-53A1349.On 737-400 s/n 25097, an undocumented repair has been found at STA867-877, STR 1-2L, under dorsal fin landing area, not conforming to anyrepair peri		- V1 Aerospace Repair R21-01067-001, Revision 02- V1 Aerospace Damage Tolerance Substantiation D21-01067-001, Revision01- FAA AMOC approval letter 792-22-00129.	(FAA) AD 2017-22-04	05/05/2023	Active
0081890		APTOZ EHF.	737-400	-All provisions of AD 2000-05-29 that are not specifically referencedabove remain fully applicable and must be compliedwith accordinglyOnly applicable to Boeing 737-400 airplane, s/n 25147, modified tofreighter in accordance with EASA STC 10015732 (FAA STC ST01827LA)Additional limitations as pe	Repetitive inspections to detect cracking of various areas of theforward pressure bulkhead<(>,<)>AD 2000-05-29 was issued to detect and correct cracking of the forwardpressure bulkhead, which could grow in size until rapiddepressurization.Cracks were found in the forward pressured deck side chord an		-V1 Aerospace Repair R22-014309-001, Revision 02-V1 Aerospace Damage Tolerance Substantiation D22-01430-001, Revision 00-FAA AMOC approval letter 792-22-00037	FAA 2000-05-29	09/05/2023	Active
0081891		APTOZ EHF.	737-400	-All provisions of AD 2017-20-14 that are not specifically referencedabove remain fully applicable and must be compliedwith accordinglyOnly applicable to Boeing 737-400 airplane, s/n 25147, modified tofreighter in accordance with EASA STC 10015732 (FAA STC ST01827LA)Additional limitations as pe	Doors, Fuselage, Nacelles/Pylons, Stabilizers, Wings - StructuralSignificant Items - Inspections / Maintenance or Inspection ProgramRevisionAD 2017-20-14 was issued to detect and repair cracking on certainstructural significant items (SSI).During a C-check on 737-400 s/n 25147, corroded area and cra		-V1 Aerospace Repair R22-01429-001, Revision 01-V1 Aerospace Damage Tolerance Substantiation D22-01429-001, Revision 00-FAA AMOC approval letter 794-22-00037	FAA 2017-20-14	09/05/2023	Active





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10081892		APTOZ EHF.	737-400	-All provisions of AD 2014-25-09 that are not specifically referencedabove remain fully applicable and must be compliedwith accordinglyOnly applicable to Boeing 737-400 airplane, s/n 25147, modified tofreighter in accordance with EASA STC 10015732 (FAA STC ST01827LA)Additional limitations as pe	Fuselage - Forward Airstair Stowage Doorway Skin Assembly and Bear Strap- InspectionAD 2014-25-09 was issued to detect and correct fatigue cracking of theskin assembly and bear strap of the aft lower corner of the forwardairstair stowage doorway, which could result in rapid loss of cabinpressure.Dur		-V1 Aerospace Repair R22-01429-001, Revision 01-V1 Aerospace Damage Tolerance Substantiation D22-01429-001, Revision 00-FAA AMOC approval letter 794-22-00037	FAA 2014-25-09	09/05/2023	Active
10081902		APTOZ EHF.	737-400	-All provisions of AD 2017-20-14 that are not specifically referencedabove remain fully applicable and must be compliedwith accordinglyOnly applicable to Boeing 737-400 airplane, s/n 25110Additional limitations as per V1 Aerospace document D21-01050-001 Rev.00, section 5, Notes	Doors, Fuselage, Nacelles/Pylons, Stabilizers, Wings - StructuralSignificant Items - Inspections / Maintenance or Inspection ProgramRevisionAD 2017-20-14 was issued to provide supplemental inspections that arerequired for timely detection of fatigue cracking for certain structuralsignificant items (-AAR Aircraft Services Repair EA 14406 Initial Revision-V1 Aerospace Damage Tolerance Substantiation D21-01050-001, Revision 00-FAA AMOC approval letter 790-21-13796	FAA 2017-20-14	10/05/2023	Active
10081903		APTOZ EHF.	737-400	-All provisions of AD 2012-18-13R1 that are not specifically referencedabove remain fully applicable and must be complied with accordinglyOnly applicable to Boeing 737-400 airplane, s/n 25110Additional limitations as per V1 Aerospace document D21-01050-001 Rev.00, section 5, Notes	Fuselage - Aft Pressure Bulkhead Web - Inspection / RepairAD 2012-18-13R1 was issued to issued to detect and correct fatiguecracks in the aft pressure bulkhead, which could resultin rapiddecompression of the fuselage.During a C-check in 2011, corrosion was found on the STA 1016 aftpressure bulkhead		-AAR Aircraft Services Repair EA 14406 Initial Revision-V1 Aerospace Damage Tolerance Substantiation D21-01050-001, Revision 00-FAA AMOC approval letter 790-21-13796	FAA 2012-18-13R1	10/05/2023	Active
10081904		APTOZ EHF.	737-400	-All provisions of AD 2016-20-02 that are not specifically referencedabove remain fully applicable and must be compliedwith accordinglyOnly applicable to Boeing 737-400 airplane, s/n 25110Additional limitations as per V1 Aerospace document D21-01050-001 Rev.00, section 5, Notes	Fuselage - Aft Pressure Bulkhead Web - InspectionAD 2016-20-02 was issued to detect and correct fatigue cracking of theaft pressure bulkhead web at the "Y"- chord, which could result inreduced structural integrity of the airplane and rapid decompression ofthe fuselage.During a C-check in 2011, corr		-AAR Aircraft Services Repair EA 14406 Initial Revision-V1 Aerospace Damage Tolerance Substantiation D21-01050-001, Revision 00-FAA AMOC approval letter 790-21-13796	FAA 2016-20-02	10/05/2023	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10074059		ASL AIRLINES FRANCE S.A.	737-300/-400/-700/-800	This AMOC only applies to the following serial numbers: 25124, 28898,29336, 24709, 25063, 25595, 29080, 29081, 32414, 32418, 32427, 30882 and30883.All provisions of AD 2014-13-07 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.	Lights – Tail Strobe Light Housing – InstallationThis AMOC provides an extension to the compliance time to paragraphs(f), (g), (h) and (i) of the AD 2014-13-07 until 15th of March2021.AD 2014-13-07 requires to install a new tail strobe light and newdisconnect bracket, and changing wire bundles. Thi		FAA AMOC approval 783-20-410.Boeing non- technical objectionin message number EUE- EUE-20-0058-03B.	US 2014-13-07	18/08/2020	Active
10078927		ASL AIRLINES (IRELAND) LIMITED	737-400	- This approval is applicable to Boeing 737-400, SN 28490 aircraft- All provisions of AD 2010-05-13 not specifically referenced in theAMOC description remain fully applicable and must be complied withaccordingly- This AMOC only applies to the AD listed above	EASA validation of the FAA AMOC ref. 790-22-150 dated 14.01.2022.Description of AMOC from the FAA AMOC approvalAD 2010-05-13 requires inspection for scribe lines and cracksin the fuselage skin at the particular locations and corrective actionas necessary in accordance with the Boeing Alert SB 737-53		FAA AMOC ref. 790-22-150 dated 14.01.2022Traces Aerospace Engineering Repair/Authorization (ERA) TS1119ER10,Revision F,dated 05.03.2022Traces Aerospace ERA TS1119ER15-A, RevisionB, dated 25.08.2016Traces Aerospace ERA TS1119ER16, Revision D, dated 05.03.2022Traces Aerospace ERA TS1119ER19, Revisio	2010-05-13	29/03/2022	Active
10072504		ATR AVIONS DE TRANSPORT REGIONALG.I.E	ATR 72-212A ATR 72-211, ATR 72-212 ATR 72-201, ATR 72-202 ATR 72-101, ATR 72-102	No flight hours accumulated with the bush P/N D61002 installed in wrongposition.Conformity restored per the advanced copy of the JIC 321100 RAI 10000provided in attachment of the AOM 2019/12 issue 02 or later	ATA 32 – Landing Gear – Main Landing Gear Bush – InspectionSeveral cases of Main Landing Gear (MLG) bush Part Number (P/N) D61002inversion have been reported from in-service aircraft. Such inversioncould had been done during any MLG RAI (Removal And Installation) and,if not detected, could result in		ATR Airworthiness Operator Message (AOM) 2019/12 Issue 2 orlater	2019-0236	12/02/2020	Active
10076311	REV. 5	ATR AVIONS DE TRANSPORT REGIONALG.I.E	ATR 72-211/-212/-212A ATR 72-101/-102/-201/-202 ATR 42-300/-320/-400/-500	None	ATA 53 – Fuselage – Passenger Seat Tracks – Inspection /Repair / ReplacementSee Annex 1 for Definition and Content.		Technical note ES-384/21 Issue 02 dated 17 March 2021Technical note ES-2020/21 Issue 03 dated 29 September 2021Seat track dynamic test report EC-1115/22 issue 01Seat track dynamictest report EC-2629/22 issue 01	2020-0097R1	16/08/2022	Active
10073988		AVIATION PARTNERS BOEINGWINGLET S COMPANY, LLC	757-200/-200PF/-300	This approval is applicable to all Boeing model 757-200/300airplanesthat have STC ST01518SE / EASA STC 10015659 winglets installed.	Time Limits / Maintenance Checks - Airworthiness LimitationsInstructions - Principal Structural Elements - Inspections / Maintenanceor Inspection Program RevisionAMOC to AD US-2020- 01-18 for airplanes with FAA STC ST01518SE / EASA STC10015659 installed.AD 2020-01-18 requires that structural inspecti		- APB document AP57.2-0604.2 (March 2020), "Supplement to D622N001 757MPD for Boeing 757-200 with winglets"- APB document AP57.3-0604.2 (March 2020), "Supplement to D622N001 757MPD for Boeing 757-300 with winglets"	US-2020-01-18	10/08/2020	Active





EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
0080495		AVIONICS INTEGRATION & ENGINEERINGCO RPORATION AG (AIEC)	CL-600-2B16 (601-3A VARIANT)	This AMOC is issued against CL-600-2B16 MSN 5042 onlyAny Avionic cockpit update/ upgrade require compliance determination iawto this AMOC	Cold Weather Operations – Introduction of Low Temperature Ground WingAnti-Ice SystemAMOC to CF-2019-04 for CL-600-2B16:-MSN 5042 only-Installation of SB 601-0641 for LTGWAIS on MSN 5042 not requiredMitigation iawAFM PSP 601A-6 supplemented by AIEC approved CL61 Checklist 0518 andQRH for MSN 5042-C		Built standard CL-600-2B16 MSN 5042 including UASC AvionicTo be operated iaw. AFM PSP 601-0641, QRH MSN 5042 and CL61 Checklist0518-AC7006a latest approved Revision by AIECWinter operation require Crew familiarization/ awareness iaw. CL61Checklist 0518, latest revision	CF-2019-04	02/11/2022	Active
0073165		BABCOCK MISSION CRITICAL SERVICESESPAN A SAU	CL-215-1A10	EASA endorses the TCCA AMOC No. AARDG 2020/ A34, submitted by letterRDIMS 16463461, issued for serial numbers 1034 (EC-HET) and 1038 (EC-HEU):Viking Air Ltd. (Viking) requested (Ref. 1) to Transport Canada CivilAviation (TCCA) for an AMOC-equivalent letter documenting ourconcurrence with the use of th	AMOC APPROVAL FOR REO 215-57- V006 REV C AND REO 215-57- V007REV DAMOC approval request to comply with inspections included on TCCA ADCF-2019-07, corrective actions,part C, accomplished according inspections included on REO 215-57- V006rev Cand REO 215-57-v007 rev D. Inspection instructions receivedV		TCCA AMOC No. AARDG 2020/A34, submitted by letter RDIMS 16463461	CF-2019-07	30/04/2020	Active
0072886		BLUEBIRD LTD	737-300, 737-400	AMOC agreement should cover only the modified area<(>,<)> thus the ADstill applies to the unmodified areas.Serial Numbers:TF-BBJ MSN 24436, TF-BBK MSN 26302, TF-BBL MSN 28887 and TF-BBM MSN25376	Background:EASA AD 2011-24-12 requires inspections as identified in the Boeing SB737-53A1301 to detect cracks on the steps of chemically milled fuselageskin, which has been modified by the STC EASA.IM.A.S.01754 (FAAST01827LA). As a result of the mentioned STC the area to be inspectedhas been modifie		SSID R-816, Revision S, dated 11.09.2018 or later approved revision.	2011-24-12	26/03/2020	Active
0073659		BLUE PANORAMA AIRLINES S.p.A.	737-800	1. All provisions of AD 2016-04-06 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.2. This approval is applicable only to Boeing Model 737-800 seriesaircraft, MSN 30007, 30289, and 30567.3. This AMOC is not transferable with the aircraft to ano	Blue Panorama Airlines B737-800 aircraft MSN 30007, 30289 and 30567 aregranted an extension of the FAA AD 2016-04-06 par. (g) and (h)compliance time up to 31 December 2020, instead of the originalcompliance date 30 June 2020.		785-20-8185: Change of Compliance Time (CCT) Alternative Method ofCompliance (AMOC) to Federal Aviation Administration (FAA) AirworthinessDirective (AD) 2016-04-06 letter, dated June 22, 2020,,Issue No. 1BPA-BPA-20-0051-06B: FAA AD 2016-04-06 Compliance Time Extension. Emailfrom The Boeing Company t	US 2016-04-06	30/06/2020	Active
0079014		CARGOLUX ITALIA S.P.A.	747-400F	Limited to 747-400F S/N 35804.	All paragraphs of the AD FAA 2003-11-01 are applicable except the dispositions as described below:1. Clean the canted pressure deck drains, as per AD 2003-11-01 § (a),with repeat interval not to exceed 24 months instead of 18 months2. Performance of work package 1 and 2 per Boeing SB 747-51A2057 asre		FAA Approval Letter Ref 120S-14-87 dated 18.02.2014.	FAA 2003-11-01	07/04/2022	Active
0077154	REV. 1	CELAIRION GmbH	A320-211	None	The Airworthiness Directive 2016-0257 mandates Airbus SB A320-71-1068accomplishment, which support the introduction of lockable fan cowllatches on CFM56 engines fitted to Airbus A320 family aeroplanes.Aircraft A321-211, MSN 6056, was modified, prior to the entry into forceof this AD, through the Air		DFP - WO 401688841DFP - WO 401688868DFP - WO 401800987DFP - WO 401801505AIRBUS TA ref: TA80068536/002/2015VSB RA32071-163.	2016-0257	05/10/2021	Active





EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10072682	REV. 1	DASSAULT AVIATION	MYSTÈRE FALCON 20	If fitted with Honeywell (formerly Allied-Signal, Garrett AirResearch) TFE731-5AR-2C or TFE731-5BR-2C turbofan engines.	Engine - First Stage Low Pressure Turbine (LPT1) Blades / VibrationMonitoringInstallation of new LPT1 and LPT2 rotor assembly blade retainers inaccordance with the accomplishment instructions of Honeywell ServiceBulletin TFE731-72-3737 rev. 2, dated 17th June 2019, constitutesterminating action forA		None	2009-0175 REV. 1	06/03/2020	Active
10075518		DASSAULT AVIATION	FALCON 7X	Applicable to all serial numbers, including those that haveembodiedM1000.	The temporary AFM modification introduced by AFM-CP0173- PUBrequiresflight crews to perform a repetitive check of flight deck oxygen masksbefore the first take-off of the day allowing for postponing thecompletion of the actions required byEASA AD 2021-0036-E, sect."Required Action(s) and Compliance		DGT105608 AFM-CP0173- PUB for Falcon 7X aeroplanesDGT147681 AFM-CP0173-PUB for Falcon 8X aeroplanes.	2021-0036-E	28/01/2021	Active
10079564	REV. 1	DEUTSCHE LUFTHANSA AG	A350-941	The AMOC is only valid for Airbus A350-941 MSN: 00136 when Airbus ASACref.: 81074747/032/2022 is complied with.Supersedure of the EASA AD 2020-0139R1 will invalidate this AMOC.	ATA 55 – Stabilizers – Horizontal Tail Plane Lateral Load FittingBushings – InspectionHORIZONTAL STABILIZER LATERAL LOAD FITTING – BUSH MIGRATIONDuring inspection according to SB A350-55-P013 required by thereferenced AD, 4ea migrated Bushes were found on all lateral loadfittings (LH Upper: 6.0 mm;I		Airbus ASAC ref.: 81074747/032/2022.	2020-0139R1	21/07/2022	Active
10079681	REV. 1	DEUTSCHE LUFTHANSA AG	A350-941	14-December-2022, OR 2428 FC of MSN 0287, whichever occurs first.	ATA 05 – Time Limits / Maintenance Checks – Damage TolerantAirworthiness Limitation Items – ALS Part 2 – AmendmentDuring accomplishment of ALI 553000-00002-01A SPECIAL DETAILEDINSPECTION (TENSION CHECK) OF VERTICAL STABILIZER TENSION STUDSFR 94 TOFR 98 during 36Mth Check on MSN0287, MP A350- A-55-3X		Airbus ASAC ref. 80999436/022/2021 issue 1, dated 14/12/2021.	2022-0125	11/07/2022	Active
10082656		DEUTSCHE LUFTHANSA AG	A340-642	- It must be ensured that the area is clean enough to detect any damage.If the area is not clean enough to detect damage, the process must bestopped and perform the required procedures in accordance with AMM57-23-00-210-818-A & 57-23-00-220-823-A A maximum distance for inspection using an endoscop	ATA 05 – Time Limits / Maintenance Checks –Damage Tolerant AirworthinessLimitation Items –ALS Part 2 – AmendmentTo perform the Detailed Visual Inspection as per the below AMM tasks,the AMM requires removal of the Right Hand & Left Hand Outer Vent Pipes (between Rib 23-35), which requires removal of t		Airbus ASAC 81260671/020/2023 issue 2, dated 05 September 2023.	2022-0188	07/09/2023	Active
10082849		DHL Air (Austria) GmbH	767-300	Limited to B767-300 msn 27686.	This AMOC is requested for the import to the Austrian register of DHLAir (Austria) GmbH with future registration of OE- LYA.AD 2016-25-01 was issued to prevent incorrect stabilizertrimadjustments during take-off, which could result in the unsafe conditionof a high speed rejected Take- off and runwayc		Kalitta Engineering Order (EO) 2211-17-02.	AD 2016-25-01	25/09/2023	Active



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0086672		DLR DEUTSCHES ZENTRUM FUERLUFT- UND RAUMFAHRT e.V.	V2527-A5	Limited to Serial Numbers: V10200 and V102011. The affectedengines do not accrue more than 100 FC per year or 200FH per year.2. The limitations and conditions contained in FAA letter reference525-25-00027, dated 11 March 2025 apply.	AD US-2015-10-04, hereafter referred as the AD, addresses corrosion andcracking of the HPC Stage 3 to 8 drums and requires among othersrepetitive inspections every 750 Flight Cycles (FC) and ultimatelyreplacement of the affected components, paragraphs (f) and (j) of the ADrefer.The applicant propose		Federal Aviation Administration (FAA) letter reference 525-25-00027< (>,<)> subject Alternative Method of Compliance to Airworthiness Directive (AD) 2015-10-04, dated 11 March 2025.	US-2015-10-04	19/03/2025	Active
0086709		DOA AESGLOBAL LIMITED	737-800	N/A	Equipment / Furnishings - Support Brackets of the Air ConditioningOutlet Extrusions between Body Station 360 and BS 907- Inspections /ModificationThe operator requires an AMOC tothe stated AD to accommodate customizedinspection instructions received from Boeing reference STA 360L and theapplicabil		- SB-737-25A1544-04 - GRP 3 (YC731 - MSN 29036)- AES-1388-01-MS- AES-1388-01-TR-01 R0 - INTRO OF AMOC FOR EASA AD US-2024-18-07Application ref. AES-1388	US-2024-18-07	18/03/2025	Active
0075128		E A D AEROSPACE AIRWORTHINESS	A330/A340 IDENTIFIED IN THE AD	Only applicable to aircraft MSNs identified in EAD Aerospace ServiceBulletin ref. 01604-IST-A-SRB, rev A.	AD 2020-0083 identifies an unsafe condition in relation with Emergencylocator Transmitter (ELT)Part Number (P/N) 01N65900 or 01N65902 installed on some Airbusproducts. The AD mandates removing the unsafe condition through theaddition of a protective diode in the installation.The AD requirement appli		- EAD Aerospace Service Bulletin ref. 01604-IST-A- SRB, rev A Modification Approval Sheet ref. 01604- EAS-A-MAS-01, RevB.	2020-0083	10/12/2020	Active
0074582	REV. 2	E A D AEROSPACE AIRWORTHINESS T/AECLIPSE TECHNICS	A319/A320/A321	Applicable MSNs are identified in the EAD Aerospace ServiceBulletinref. 01604- IST-A-SRB.	AD 2020-0103 identifies an unsafe condition in relation with Emergencylocator Transmitter (ELT) Part Number (P/N) 01N65900 installed on someAirbus Single-Aisle aircraft. The AD mandates removing the unsafecondition through the addition of a protective diode in theinstallation.EAD Aerospace aircrafts		- EAD Aerospace Service Bulletin ref. 01604-IST-A- SRB, rev C Modification Approval Sheet ref. 01604- EAS-A-MAS-01, RevB.	2020-0103	12/12/2023	Active
0072178	REV. 1	EASYJET AIRLINE COMPANY LIMITED	A319-111, A320-214 A319-111 / A320-214	This AMOC applicability is limited to aeroplanes on which either Airbusmodification 157519 has been embodied in production or Airbus SBA320-71-1068 has been embodied in service< (>,<)> and which are in"config 004" (A320) or "config 006" (A319), as defined through this SB.This AMOC is limited to Seria	This AD mandates Airbus SB A320-71-1068 introducing lockable fan cowllatches on CFM56 engines fitted to Airbus A320 family aeroplanes andincludes stowing the keys of the fan cowl latches in the box at thebottom of the coat stowage, as applicable to aeroplane configuration.easyJet had initially stowe		Airbus A320 IPC 25-13-01-29MDesign Change Document EZE-1470M Issue 03(The use of later approved revisions of this document is acceptable.)	2016-0257	05/02/2021	Active
0080072		EASYJET AIRLINE COMPANY LIMITED	A319-111, A320-214	This AMOC applicability is limited to aeroplanes on which either Airbusmodification 157519 has been embodied in production or Airbus SBA320-71-1068 has been embodied in service, and which are in "config 004"(A320) or "config 006" (A319), as defined through this SB.This AMOC replaces both AMOC approv	This AD mandates Airbus SB A320-71-1068 introducing lockable fan cowllatches on CFM56 engines fitted to Airbus A320 family aeroplanes andincludes stowing the keys of the fan cowl latches in the box at thebottom of the coat stowage, as applicable to aeroplane configuration.easyJet had initially stowe		Airbus A320 IPC 25-13-01-29MDesign Change Document EZE-1470M Issue 03(The use of later approved revisions of this document is acceptable.)	2016-0257	07/09/2022	Active





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0084317	REV. 1	EASYJET AIRLINE COMPANY LIMITED	A319-111 AND A320-214	This AMOC applicability is limited to aeroplanes on which either Airbusmodification 157519 has been embodied in production or Airbus SBA320-71-1068 has been embodied in service, and which are in "config 004" (A320) or "config 006" (A319), as defined through this SB.This AMOC replaces AMOC approval 10	This AD mandates Airbus SB A320-71-1068 introducing lockable fan cowllatches on CFM56 engines fitted to Airbus A320 family aeroplanes andincludes stowing the keys of the fan cowl latches in the box at thebottom of the coat stowage, as applicable to aeroplane configuration.easyJet had initially stowe		Airbus A320 IPC ref 25-13-01-29 / 29L item 10Design Change Document EZE-1470M Issue 5(The use of later approved revisions of this document is acceptable.)	EASA 2016-0257	10/05/2024	Active
10073569	REV. 1	EIRTECH AVIATION SERVICES LIMITED	737-900/-900ER 737-600/-700/-700C/-800	None	Equipment / Furnishings - Passenger Service Units and Life Vest Panels - ModificationThis AMOC to US AD 2019-03-26 (1). is based on an existing Minor Change(2) issued by Eirtech Aviation Services (under their EASA privileges)for installation of PSU and Life Vest Panel Lanyard Assemblies on 737-600,		MAC/1368 - Approval Certificate: Lanyard Replacement - Passenger ServiceUnit and Life Vest Panel - 737 - Issue 1MDL/ 1368-001 - Master Document List: Lanyard Replacement - PassengerService Unit and Life Vest Panel - 737 - Issue 1AMS/1368-001 - Aircraft Manual Supplement: 737 Lanyard Replacement –Pass	US-2019-03-26	23/06/2020	Active
10079200		EIRTECH AVIATION SERVICES LIMITED	737-800/-900ER 737-600/-700/-700C		This AMOC covers the US AD 2021-26-17, which supersedes AD 2019-03-26,already covered by EASA AMOC approval Ref. 10073569.The new AD 2021-26-17 retains the requirements of AD 2019-03-26, and,for certain airplanes, requires an inspection todetermine if there-identified PSU part number is correct, an		MAC/1368 - Approval Certificate: Lanyard Replacement - Passenger ServiceUnit and Life Vest Panel - B737 Issue 3,,MDL/ 1368-001 - Master Document List: Lanyard Replacement - PassengerService Unit and Life Vest Panel - B737 Issue 3AMS/1368-001 - Aircraft Manual Supplement: B737 Lanyard Replacement -Pas	US 2021-26-17	06/05/2022	Active
10072651	REV. 1	ELBE FLUGZEUGWERK E GmbH	A330-323/-341/-342/-343 A330-301/-302/-303/-321/- 322 A330-201/-202/-203/-223/- 243	The AMOC is only valid for the Airbus A330-200 and A330-300airplanemodels modified in accordance with EASA STC 10063798.Supersedure of the EASA AD 2018-0276R1 may invalidate thisAMOC.	Fuselage - Structural Parts/ Joints - Modification / ReinforcementThe AD and the associated Service Bulletins are partially not applicableto A330-300 and A330-200 airplanes which have undergone the passenger tofreighter (P2F) conversion prescribed by the EASA STC Project Number0010042837 (for A330-3		EASA STC 10063798.398/ AMOC/032 Rev. C: "Alternative Methodsof Compliance forAirworthiness Directive (AD) 2018-0276R1",dated 17 May 2021TA3980-08SA, Fatigue and Damage Tolerance Analysis A330 P2F Center WingBox, ST AerospaceSTR029F, A330-300 P2F WFD Affected Areas, ST AerospaceSTR051H, A330-200 P2F	2018-0276R1	24/01/2024	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
0072652		ELBE FLUGZEUGWERK E GmbH	A330-323/-341/-342/-343 A330-301/-302/-303/-321/- 322 A330-201/-202/-203/-223/- 243	The AMOC is proposed to be applicable to A330-300 airplaneswhich have undergone the passenger to freighterconversion prescribed bythe EASA STC 10063798.On top of thesupplementary maintenance tasks specified in A330P2F/ ALSPart 2 supplement Rev 06, the maintenance tasks specified in Air	AD 2019-0059 mandates compliance with the instructions contained inAirbus A330 Airworthiness Limitations Section (ALS) Part 2 Revision 03.EFW proposes that for A330-300 and A330-200 passenger aeroplanesconverted to freighter configuration as per STC 0010063798, theimplementation of ALS Part 2 Supple		398/AMOC/038 Rev. B: "Alternative Methods of Compliance forAirworthiness Directive (AD) 2019-0059"A330P2F/ ALS Part 2 Supplement Rev 06.	2019-0059	30/03/2020	Active
0073521		ELBE FLUGZEUGWERK E GmbH	A330-323/-341/-342/-343 A330-301/-302/-303/-321/- 322 A330-201/-202/-203/-223/- 243	The AMOC is only valid for the Airbus A330-200 and A330-300airplanemodels modified in accordance with EASA STC 10063798.Supersedure of the EASA AD 2019-0047, or the update of Airbus A330 ALSPart 4 Revision 07 as well as the update of the ST Aerospace STC ALSPart 4 Supplement may invalidate this AMO	The EASA AD 2019-0047 requires to comply with the Airbus A330 ALS Part 4'System Equipment Maintenance Requirements (SEMR)' at Revision 07.ST Engineering Aerospace Pax to Freighter (P2F) conversion STC (EASAapproval 10063798), applicable tothe Airbus A330-200 /-300 aircraftmodels, includes through i		EASA STC 10063798.A330_ALS_Part 4_SEMR_R07A330P2F_AL S_Part4-SEMR-ENV SEMR Supplement ENV R00_Apr_01_20A330P2F/ ALS_Part4-SEMR-ENV at R00398/AED/ICA at Rev.E	2019-0047	24/06/2020	Active
0073995		ELBE FLUGZEUGWERK E GmbH	A321-211/-231	The AMOC is only valid for the Airbus A321-200 aircraft models modifiedto A321P2F in accordance with EASA STC 10071994.	Fuselage – Frame 35.2A - InspectionThe EASA AD 2016-0106R1 requires repetitive special detailed (rototest) inspections (SDI) of the affected holes and, depending on findings,accomplishment of a repair.With the Elbe Flugzzeugwerke GmbH (EFW) Pax to Freighter (P2F)conversion STC (EASA approval 10071994		EASA STC 10071994Airbus SB A320-53-1355 (including roto probe inspection, cold workingand fastener installation)349/AMOC/007 Rev. B Alternative Method of Compliance for AirworthinessDirectives EASA AD 2016-0106R1 and FAA AD 2017-10-23	2016-0106R1	04/08/2021	Active
0073996		ELBE FLUGZEUGWERK E GmbH	A321-211/-231	The AMOC is only valid for the Airbus A321-200 airplane models modifiedin accordance with EASA STC 10071994.	Fuselage – Central Vertical Strut at Frame 65 – ModificationThe EASA AD 2016-0212 requires a modification of the Central VerticalStrut at Frame 65 in order it cannot penetrate through the cabin flooras a consequence of an airframe ground contact above certified verticalspeed/vertical acceleration.To		EASA STC 10071994349/ AED/STR039 Static Substantiation of Vertical Struts for 14P349/AED/008 Document Master List for A321P2FAirbus SB A320-53-1262 or SB A320-53-1334349/AMOC/ 012 Rev. B Alternative Method of Compliance for AirworthinessDirectives EASA AD 2016-0212 and FAA AD 2017-25-06	AD 2016-0212	04/08/2021	Active
0074009		ELBE FLUGZEUGWERK E GmbH	A321-211/-231	The AMOC is only valid for the Airbus A321-200 aircraft models modifiedto A321P2F in accordance with EASA STC 10071994.	Fuselage – Crossbeam Splicing at Frames 16 and 20 – InspectionThe EASA AD 2016-0139 repetitive special detailed inspections (SDI) ofthe two upper rows of fasteners of the crossbeam splicing at FR16 andFR20, on both LH and RH sides, and, depending on aeroplaneconfiguration, provides an optional termi		EASA STC 10071994AIRBUS SB A320-53-1286 Fuselage – Nose Forward Fuselage – Inspection atCrossbeam Splicing FR16/20 LHS/ RHSAIRBUS SB A320-53-1295 Fuselage – Main Structure – Introduce Bolt withCold Working and Moderate Interference on Crossbeam Splicing FR 16 andFR20349/AED/ STR050C Fatigue and Damage	2016-0139	04/08/2021	Active





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10074010		ELBE FLUGZEUGWERK E GmbH	A321-231	The AMOC is only valid for the Airbus A321-231 airplane models modifiedin accordance with EASA STC 10071994.	ATA 53 – Fuselage – Frame 35 / Slidebox Junction - InspectionThe EASA AD 2016-0146R1 mandates the accomplishment of an SDI at thelocations of FR 35 as specified in Table 2 of AD 2016-0146R1 and inaccordance with the instructions of the applicable Airbus SB as definedin Table 2 AD 2016-0146R1within t		EASA STC 10071994AIRBUS SB A320-53-1308 – Fuselage – Forward Fuselage – Holes InspectionFrames 35.1 LHS Under SlideboxAIRBUS SB A320-53-1309 – Fuselage – Forward Fuselage – Holes InspectionFrames 35.1 RHS Under SlideboxAIRBUS SB A320-53-1310 – Fuselage – Forward Fuselage – Holes InspectionFrames 35	2016-0146R1	18/08/2020	Active
10074122	REV. 2	ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231	The AMOC is only valid for the Airbus A321-211, A321-231 airplane modelsmodified in accordance with EASA STC 10071994.	The EASA AD 2018-0233R1 mandates the accomplishment of the repetitiveinspections, via rototest, of open tack holes and rivet holes at thecargo floor support fittings between FR 50and FR 63, between Stringer(STGR) 33 and STGR 39 (LH side only), in accordance with theinstructions of the inspection SB		EASA STC 10071994Airbus SB A320-53-1257 Fuselage – Inspection of Tack Holes at FR 50 – FR63 between P33 – P 39 (LH Only) in Cargo Floor AreaAirbus SB A320-53-1261 Fuselage – Rear Fuselage – Reinforce Frames inArea of P31 – P39 LHS Section 17Airbus SB A320-53-1360 Fuselage – Rear Fuselage – Reinforce	2018-0233R1	07/07/2021	Active
10074583		ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231	The AMOC is only valid for the Airbus A321-211, A321-231 airplane modelsmodified in accordance with EASA STC 10071994.	EASA AD 2020-0036R1 mandates to apply Airbus ALS Part 2 Revision 08, andto update the operator Aircraft Maintenance Programme (AMP) within 12months, from the effective date of theoriginal issue of this AD [i.e.21 March 2020].As part of their passenger to freighter (P2F) conversion STC, referencedEA		EASA STC 10071994AIRBUS A318/ A319/A320/A321 Airworthiness Limitations Section (ALS) Part2 Revision 07, dated 13 Jun 2018AIRBUS A318/A319/ A320/A321 Airworthiness Limitations Section (ALS) Part2 Revision 08, dated 11 Oct 2019EFW Airworthiness Limitation Section ALS Part 2 Airworthiness LimitationItems	2020-0036R1	09/03/2021	Active
10074812		ELBE FLUGZEUGWERK E GmbH	A321-231 A321-211	The AMOC is only valid for the Airbus A321-211, A321-231 airplane modelsmodified in accordance with EASA STC 10071994.	In the frame of the A321 Passenger to Freighter conversion (A321P2F),the Passenger Cabin was replaced by a Main Deck Cargo Compartment.Consequently the passenger oxygen system wasdeleted and the smokedetection system was replaced by a newdedicated one.The AD mandates the accomplishment of all main		- EASA STC 10071994- Airbus A318/A319/A320/ A321 ALS Part 3 Revision 7 dated 31 Jul 2020."Airbus A318/A319/A320/A321 Airworthiness Limitations Section (ALS) Part3 – Certification Maintenance Requirement (CMR)"- A320/A321 P2F ALS Part 3 Supplement Envelope (ENV) Revision 00 dated01 Aug 2019. "A320/A32	2020-0067	11/11/2020	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10074978		ELBE FLUGZEUGWERK E GmbH	A330-341/-342/-343 A330-321/-322/-323 A330-301/-302/-303 A330-201/-202/-203/-223/- 243	The AMOC is only valid for the Airbus A330-200 and A330-300airplanemodels modified in accordance with EASA STC 10063798, and are embodiedwith Service Bulletin A330-21-3053, A330-92-3017 and A330-34-3044, andwhich are not embodied with Airbus Modifications 44457 and 45022.Supersedure of the EASA AD	The EASA AD 97-154-049R1 was issued to protect against presence ofmoisture inside the rear electrical connectors for THOMSON radioaltimeter transceivers which can cause incorrect fluctuations of theradio height and therefore reduce the reliability of these radioaltimeters.Mandatory embodiment of the		EASA STC 10063798.SB A330-21-3053 R01SB A330-92-3017 R04SB A330-34-3044 R01Airbus MOD-44457Airbus MOD-45022	97-154-049(B)	26/11/2020	Active
10074985		ELBE FLUGZEUGWERK E GmbH	ALL APPROVED A330 MODELS A330-323/-341/-342/-343 A330-301/-302/-303/-321/- 322 A330-201/-202/-203/-223/- 243	The AMOC is only valid for the Airbus A330-200 and A330-300airplanemodels modified in accordance with EASA STC 10063798 and which areitemized in SB A330-28-A3060 that are withoutembodiment of MOD 47293 orSB A330-28-3063.Supersedu re of the AD 1999-046-091R4 may invalidate this AMOC.	The AD 1999-046-091R4 mandates detailed visual inspections of the trimfuel line between FR77 and FR86 at intervals not exceeding 1000 flighthours, in accordance with instructions given in SB A330–28A3060.This is to ensure that the trim fuel line and shrouds are not damageddue to excessive pressure b		EASA STC 10063798.Airbus Service Bulletin A330-28A3060Airbus Service Bulletin A330-28-3063Airbus Mod 47293H14074EFW- SB-28-0002.	1999-046-091R4	25/01/2023	Active
10077657		ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231	The AMOC is only valid for the Airbus A321-211, A321-231 airplane modelsmodified in accordance with EASA STC 10071994.	In the frame of the A321 Passenger to Freighter conversion (A321P2F), the used circuit breaker 15WN has been removed during P2F conversion. Therefore, an alternative circuit breakerneeds to be defined. Thiscircuit breaker definition had been covered with EFW-SB-31-0003. To correct the unsafe condition		- EASA STC 10071994- EFW-E-AMOC-00004 Rev. A- Service Bulletin No.: EFW-SB-31-0003 Rev.: 01.	2017-0237	10/11/2021	Active
10077937		ELBE FLUGZEUGWERK E GmbH	A321-231 A321-211	The AMOC is only valid for the Airbus A321-211, A321-231 airplane modelsmodified in accordance with EASA STC 10071994.	The EASA AD 2021-0227 requires repetitive SDI of the affected areas and,depending on findings, accomplishment of applicable correctiveaction(s). This AD also includes reference to an optional terminatingaction for the repetitive SDI.As part of the Elbe Flugzeugwerke GmbH (EFW) A321P2F passenger tofr		- EASA STC 10071994- EFW-E-AMOC-00005 Rev. AAirbus SB A320-53-1373 original issue dated 14 June 2018.Airbus SB A320-53-1374 original issue dated 14 June 2018.Airbus SB A320-53-1378 original issue dated 14 June 2018, or Revision 01dated 17 September 2019.Airbus SB A320-53-1379 original issue dated 14	2021-0227	17/12/2021	Active
10078170		ELBE FLUGZEUGWERK E GmbH	A320-231/-232/-233 A320-211/-212/-214/-215/- 216	The AMOC is only valid for the Airbus A320-200 airplane models modifiedin accordance withEASA STC 10071994.	EASA AD 2020-0036R1 mandates to apply Airbus ALS Part 2 Revision 08, andto update the operator Aircraft Maintenance Programme (AMP) within 12months, from the effective date of theoriginal issue of this AD[i.e. 21 March 2020]. Operators are to contact Airbus in case of findingdiscrepancies during th		EASA STC 10071994349/ AMOC/022-E Rev A "Alternative Method of Compliance for AirworthinessDirectivesEA SA AD 2020-0036R1"AIRBUS A318/A319/A320/A321 Airworthiness Limitations Section(ALS) Part2 Revision 08, Issue 02,dated 19 May 2020A320/ A321P2F-ALS_Part2-ALI- ENV Rev 03 "Airworthiness Limitation Secti	2020-0036R1	20/01/2022	Active





EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10078226		ELBE FLUGZEUGWERK E GmbH	A321-211/-231 A320-231/-232/-233 A320-211/-212 /-214/-215 /-216	The AMOC is only valid for the Airbus A320-200 & A321-200 airplanemodels modified in accordance with EASA STC 10071994.	Fuselage – Crossbeam Splicing at Frames 16 and 20 – InspectionThe EASA AD 2016-0139 repetitive special detailed inspections (SDI) ofthe two upper rows of fasteners of the crossbeam splicing at FR16 andFR20, on both LH and RH sides, and, depending on aeroplaneconfiguration, provides an optional termi	Alternative Method of Compliance to Airworthiness Directive2016-0139Fuselage – Crossbeam Splicing at Frames 16 and 20 –Inspection	EASA STC 10071994A320/ A321P2F-ALS_Part2-ALI- ENV Rev 03 "Airworthiness Limitation SectionALS Part 2 Airworthiness Limitation Items ALI Supplement Envelope (ENV)"349/AED/ STR091C RevA "Fatigue and Damage Tolerance Analysis of FloorStructure (FWD Fuselage) for A320 P2F (11P)"349/AED/ STR091L-2 Rev A "Fa	2016-0139	25/07/2022	Active
10078227		ELBE FLUGZEUGWERK E GmbH	A321-211/-231 A320-231/-232/-233 A320-211/-212 /-214/-215 /-216	None	Fuselage – Cargo Compartment Fitting Brackets, Tack and Rivet Holes – Inspection / Repair. The EASA AD 2018-0233R1 mandates the accomplishment of the repetitiveinspections, via rototest, of open tack holes and rivet holes at thecargo floor support fittings between FR 50 and FR 63, between Stringer(STG	Alternative Method of Compliance to Airworthiness Directive2018-0233R1	EASA STC 10071994Airbus SB A320-53-1257 Fuselage – Inspection of Tack Holes at FR 50 – FR63 between P33 – P 39 (LH Only) in Cargo Floor AreaAirbus SB A320-53-1261 Fuselage – Rear Fuselage – Reinforce Frames inArea of P31 – P39 LHS Section 17Airbus SB A320-53-1360 Fuselage – Rear Fuselage – Reinforce	2018-0233R1	26/07/2022	Active
10078279		ELBE FLUGZEUGWERK E GmbH	A321-231/-232 A321-211/-212/-213 A320-232	The AMOC is only valid for the Airbus A320-200 and A321-200airplanemodels modified in accordance with EASA STC 10071994.	FIRST TECHNOLOGY FIRE & SAFETY - Toilet Compartment Fire ExtinguishersAlternative Method of Compliance to AirworthinessDirective EASA ADG-007-11-97The EASA AD G-007-11 97 mandates inspection and replacement required ifnecessary for affected fire extinguishers with part number (PN) BA21758,to ensure		EASA STC 10071994349/ AMOC/005 Rev C, 'Alternative Methods of Compliance for AirworthinessDirectives (AD) G-007-11-97'SB26-110, 'First Technology Fire and Safety SB 26-110'SB A320-26-1050, 'Fire Protection –Lavatory Fire Extinguishing –Do aDetailed Visual Inspection of the Lavatory Waste-Bin Fire Ex	G-007-11-97	26/10/2023	Active
10078325		ELBE FLUGZEUGWERK E GmbH	A321-211/-231 A320-231/-232/-233 A320-211/-212/-214/-215/- 216	Modified in accordance with EASA STC 10071994.	Toilet rinse valve / MONOGRAM rinse valve PN 15800-348 revision CThe EASA AD F-1997-269 mandates the replacement of theMonogram rinsevalve PN 15800-348 rev. C on A320 and A321 aircraft models not fittedwith AIRBUS INDUSTRIE modification 26145 or without AIRBUS INDUSTRIEService Bulletin A320-38-1049		EASA STC 10071994349/ AMOC/003 Rev C	AD-F-1997-269	22/07/2022	Active
10079746	REV. 1	ELBE FLUGZEUGWERK E GmbH	A321-211/-231 A320-231/-232/-233 A320-211/-212/-214/-215/- 216	All manufacturer serial numbers modified in accordance withEASA STC10071994, except aeroplanes on which Airbus modification (mod) 157039has been embodied in production.	ATA 53 – Fuselage – Door Stop Fitting Holes at Frame 66/68 – Inspection/ RepairThe EASA AD 2021-242 mandates the accomplishment of an optimizedinspection regime (threshold and interval) for the LH/RH Frames 66 and68 web, and necessary actions depending on the findings.With the passenger to freighter		EASA STC 10071994.EFW-E- AMOC-00006: "Alternative Methods ofCompliance for AirworthinessDirective (AD) 2021-0242", dated 08 Nov. 2021Drawing D534R1555 Installation, Door 4, Structure and Skin	2021-0242	12/12/2023	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10079830		ELBE FLUGZEUGWERK E GmbH	A321-211/-231 A320-231/-232/-233 A320-211/-212/-214/-215/- 216	None	ATA 53 – Fuselage – Service Door Stop Fitting Hole – Inspection / RepairEASA AD 2018 0289 R1 requires repetitive inspections of the affectedareas and, depending on findings, accomplishment of applicablecorrective action(s) That AD also included reference to the applicablemodification SB which provid		EASA STC 10071994349/ AMOC/017-E Rev A "AMOC FOR AD 2018-0289R1"	2018-0289R1	28/07/2022	Active
10079832		ELBE FLUGZEUGWERK E GmbH	A321-211/-231 A320-231/-232/-233 A320-211/-212/-214/-215/- 216	None	ATA 53 – Fuselage – Double Joggle Area Frame 16 and Frame 20 –Inspection / ModificationThe EASA AD 2021-0227 mandates repetitive Special Detailed Inspection(SDI) of the Holes in the Double Joggle Area at Frame 16 and Frame 20and, dependingon findings, accomplishment of applicable correctiveaction(s		EASA STC 10071994349/ AMOC/023-EAIRBUS SB A320-53-1373 FUSELAGE - NOSE FORWARD FUSELAGE - INSPECTIONDOUBLE JOGGLE AREA FR16AIRBUS SB A320-53-1374 FUSELAGE - NOSE FORWARD FUSELAGE - INSPECTIONDOUBLE JOGGLE AREA FR20AIRBUS SB A320-53-1378 FUSELAGE - NOSE FORWARD FUSELAGE - REDESIGN OFTHE DOUBLE JOGGLEA	2021-0227	28/07/2022	Active
10080421		ELBE FLUGZEUGWERK E GmbH	A330-341/342/343 A330-321/-322/-323 A330-301/-302/-303 A330-223/-243 A330-201/-202/-203	This AMOC is applicable for A/C A330-200 and -300 with P2F conversion.	ATA 05 – Time Limits / Maintenance Checks – Airworthiness LimitationsSectionPart 3 – Certification Maintenance Requirements – ImplementationThe EASA AD 2021-0248 requires to comply with the Airbus A330 ALS Part 3'Certification Maintenance Requirements (CMR) Rev.07The AD 2021-0248 supersedes AD 2019		EFW Statement for Alternative Method of Compliance Ref. EFW-E- AMOC-00007Rev A dated 17.10	2021-0248	20/10/2022	Active
10080424		ELBE FLUGZEUGWERK E GmbH	A330-341/342/343 A330-321/-322/-323 A330-301/-302/-303 A330-223/-243 A330-201/-202/-203	This AMOC is applicable for A/C A330-200 and -300 with P2F conversion.	ATA 05 - Time Limits / Maintenance Checks – System Equipment MaintenanceRequirements –ALS Part 4 – AmendmentThe EASA AD2021-0250 requires to comply with the Airbus A330 ALS Part 4'System Equipment Maintenance Requirements (SEMR) Rev. 08.The AD 2021-0250 supersedes AD 2019-0047 and AMOC - Doc No.:398		EFW Statement for Alternative Method of Compliance Ref. EFW-E- AMOC-00008Rev A dated 17.10.2022	2021-0250	21/10/2022	Active
10080970		ELBE FLUGZEUGWERK E GmbH	A330-300P2F A330-200P2F	The AMOC is valid in conjunction with the summary of investigationpresented with the EFW Document Ref. EFW-E- AMOC-00009Rev A dated13.12.2022The AMOC issue is based on the AD 2022-0187 provision AMP Revision (3).	With the publication of AD 2022-0187 the accomplishmentof theactions specified in the Airworthiness LimitationsSection (ALS)Airbus A330 ALS Part 2 Rev. 05 is required.The AD 2022-0187 retains the requirements of EASA AD 2021-0261,which is superseded, and requires also accomplishment ofs		EFW Statement for Alternative Method of Compliance Ref. EFW-E- AMOC-00009Rev A dated 13.12.2022.	2022-0187	21/12/2022	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10081122	REV. 1	ELBE FLUGZEUGWERK E GmbH	A321-211/-231 A320-232	Applicable to A320-232, A321-211/-231 all manufacturer serial numbersmodified in accordance with EASA STC 10071994.	Time Limits / Maintenance Checks – Airworthiness Limitations SectionPart 4 – System Equipment Maintenance Requirements – AmendmentIn the frame of the A320/A321 Passenger to Freighter conversion(A320P2F, A321P2F), the existing SEMRs definedin AirbusA318/ A319/A320/A321 ALS Part 4 can no longer be car		EASA STC 10071994349/ AMOC/026 Rev AEASA AD 2022-0102 - ATA 05 – Time Limits/Maintenance Checks –Airworthiness Limitations Section Part 4 – System Equipment MaintenanceRequirements – AmendmentAirbus A318/ A319/A320/A321 ALS Part 4 Revision 8Issue 02 dated 18February 2022 - Airbus A318/A319/A320/A321S	EASA 2022-0102	09/11/2023	Active
10081143		ELBE FLUGZEUGWERK E GmbH	A330-342, A330-343 A330-323/-341/-342/-343 A330-322, A330-323, A330-341, A330-302, A330-303, A330-321, A330-301/-302/-303/-321/- 322 A330-223, A330-243, A330-201, A330-202, A330-203, A330-201/-202/-203/-223/- 243	The AMOC is only valid for the Airbus A330-200 and A330-300airplanemodels modified in accordance with EASA STC 10063798.	The AD 2012-0090 was issued to ensure sufficient clearance betweenbonding lead from the isolation valve 283HN and electrical harness5871VB at zone 160 as per SB A330-21-3165. Thisis to prevent any riskof chaffing and possible short circuit of the isolation valve andconsequential non-closure of thep		EASA STC 10063798.Airbus Service Bulletin SB A330-21-3165EA-398-21-0 116-1385EA-398-38-0600- 0982EA-398-38-0000-0204 EA-398-38-0610-0333EFW -SB-38-0002.	2012-0090	27/01/2023	Active
10081197		ELBE FLUGZEUGWERK E GmbH	A321-251N/-253N/-271N/- 272N A321-211/-212/-213/-231/- 232 A320-271N A320-214/-216/-232/-233/- 251N A319-133 A319-111/-112/-115/-131/- 132 A318-112/-121/-122	All serial numbers modified per EASA STC 10071994, as defined throughEFW document Ref. EFW-E-AMOC-00010 Rev A.	The EASA AD 2019-0069 requires to modify the interconnecting brackets ofcertain overhead stowage compartments (OHSC) and pivoting OHSC (POHSC)to improve their robustness.The definition of the EFW A318/A319/A320/A321 Passenger-to- Freighter(P2F) conversion STC (EASA STC 10071994) removes OHSC and POHS		EFW Statement for Alternative Method of Compliance Ref. EFW-E- AMOC-00010Rev A.	2019-0069	09/02/2023	Active
10081552		ELBE FLUGZEUGWERK E GmbH	A321-211/-231 A320-231/-232/-233 A320-211/-212/-214/-215/- 216	None	ATA 53 – Fuselage – Inner Cap and Frame Flange at Frame 68 Stringer 22 - InspectionThe EASA AD 2022-0030 required repetitive special detailed inspection(SDI) of the dedicated areasand, depending on findings, accomplishmentof applicable corrective action(s).As part of the Elbe Flugzeugwerke GmbH (EFW		EASA STC 10071994EFW- E-AMOC-00011 Rev. AAirbus SB A320-53-1491	2022-0030	23/03/2023	Active
10082068		ELBE FLUGZEUGWERK E GmbH	A321-211/-212/-213/-231/- 232 A320-232		The EASA AD 2022-0091 provides requirements for the accomplishment ofthe actions specified in the Airbus Certification MaintenanceRequirements (CMR) as specified in ALS Part 3.As part of the Elbe Flugzeugwerke GmbH (EFW) A320/A321 passenger tofreighter (P2F) conversion<(>,<)> STC<(>,<)> EASA approva		EASA STC 10071994EFW- E-AMOC-00014 Rev. AAIRBUS A320FAM_ALS_Part3_CM R Revision 08 Issue 02 dated 18 February 2022EFW Doc. A320/ A321P2FALS_PART3- CMR-ENV Supplement Rev 01.	EASA AD 2022-0091	05/06/2023	Active
10082069		ELBE FLUGZEUGWERK E GmbH	A321-211/-212/-213/-231/- 232 A320-232		The EASA AD 2022-0085 provides requirements for the accomplishment ofthe actions specified in the Airworthiness Limitations Section (ALS) AIRBUS A320FAM ALS Part 2 Rev. 09.As part of the Elbe Flugzeugwerke GmbH (EFW) A320/A321 passenger tofreighter (P2F) conversion, STC, EASA approval 10071994, a sup		EASA STC 10071994EFW- E-AMOC-00012 Rev. BAIRBUS A318/A319/ A320/A321 ALS Part 2 Rev. 09 dated 02 February 2022EFW A320/A321P2F- ALS_Part2-ALI-ENV Rev 05.	EASA AD 2022-0085	05/06/2023	Active





EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10084655		ELBE FLUGZEUGWERK E GmbH	A321-231 A321-211 A320-232	None	Fuselage – Door Stop Fitting Holes at Frame 66/68 – Inspection / RepairAD 2021-0242 is fully terminated by EASA STC 10071994 Rev. 25 or laterand equivalent level of safety as imposed by EASA AD 2021-0242 isachieved for A321P2F.If, any crack or damage in the AD affected area is found, before nextflig		EASA STC 10071994EFW- E-AMOC-0006 Rev. D – Statement for Alternative Method of ComplianceA320/ A321P2F-ALS_Part2-ALI- ENV – Airworthiness Limitation Section (ALS) Part 2 AirworthinessLimitation Items ALI Supplement Envelope (ENV), Rev07 or later revision	2021-0242	07/06/2024	Active
10084656	REV. 1	ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231 A320-214, A320-232	None	Fuselage – Centre Fuselage Forward Pressure Bulkhead – ReinforcementAD 2014-0209 is fully terminated by EASA STC 10071994 and equivalentlevel of safety as imposed by EASA AD 2014-0209 is achieved.If, during any inspection as defined in theP2F ALS Part 2 Supplement orthe installation of EFW- SB-02-00		EASA STC 10071994EFW- E-AMOC-00022 Rev. A – Statement for Alternative Method of ComplianceA320/ A321P2F-ALS_Part2-ALI- ENV– Airworthiness Limitation Section (ALS) Part 2 Airworthiness Limitation Items ALI Supplement Envelope (ENV), Rev05 or later revision	EASA 2014-0209	16/08/2024	Active
10084660	REV. 1	ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231 A320-214, A320-232	None	Fuselage – Service Door Stop Fitting Hole – Inspection / RepAD 2018-0289R1 is fully terminated by EASA STC 10071994 andequivalentlevel of safety as imposed by EASA AD 2018-0289R1is achieved.If, any crack or damage in the AD affected areais found, before nextflight contact STC holder for approveda		EASA STC 10071994EFW- E-AMOC-00023 Rev. A – Statement for Alternative Method of Compliance	EASA 2018-0289R1	16/08/2024	Active
10084661		ELBE FLUGZEUGWERK E GmbH	A321-231	None	Fuselage - Frame 35 / Slidebox Junction – InspectionAD 2016-0146R2 is partly terminated by EASA STC 10071994 Rev. 25 orlater.AD 2016-0146R2 is fully terminated for the LHS for allA321P2F aircraft.AD 2016-0146R2 is fully terminated for theRHS for A321P2F aircraftPre-mod 155607.The above described a		EASA STC 10071994EFW- E-AMOC-00025 – Statement for Alternative Method of Compliance	2016-0146R2	11/06/2024	Active
10084663	REV. 1	ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231 A320-214, A320-232	None	ATA 53 – Fuselage – Crossbeam Splicing at Frames 16 and 20 – InspectionAD 2023-0150 is fully terminated by EASA STC 10071994 andequivalent level of safety as imposed by EASA AD 2023-0150 is achieved.If, during any inspection as defined in the P2F ALS Part 2 Supplement, any crack or damage in the AD a		EASA STC 10071994EFW- E-AMOC-00015 Rev. D – Statement for Alternative Method of ComplianceA320/ A321P2F-ALS_Part2-ALI- ENV– Airworthiness Limitation Section (ALS) Part 2 Airworthiness Limitation Items ALI Supplement Envelope (ENV), Rev07 or later revision	EASA 2023-0150	16/08/2024	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
0084664	REV. 1	ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231 A320-214, A320-232	None	Fuselage – Cargo Compartment Fitting Brackets, Tack and Rivet Holes – Inspection / RepAD 2022-0115 is fully terminated by EASA STC 10071994 and equivalentlevel of safety as imposedby EASA AD 2022-0115 is achieved.lf, during any inspection as defined in the P2F ALS Part 2 Supplement orthe installatio		EASA STC 10071994EFW- E-AMOC-00018 Rev. C – Statement for Alternative Method of ComplianceA320/ A321P2F-ALS_Part2-ALI- ENV– Airworthiness Limitation Section (ALS) Part 2 Airworthiness Limitation Items ALI Supplement Envelope (ENV), Rev05 or later revision	EASA 2022-0115	16/08/2024	Active
0084665	REV. 1	ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231 A320-214, A320-232	None	Fuselage – Double Joggle Area at Frame 16 and Frame 20 – InspectionAD 2023-0212 is fully terminated by EASA STC 10071994 and equivalentlevel of safety as imposed by EASA AD 2023-0212 is achieved.If, during any inspection as defined in the P2F ALS Part 2 Supplement,any crack or damage in the AD affec		EASA STC 10071994EFW- E-AMOC-00021 Rev. A – Statement for Alternative Method of ComplianceA320/ A321P2F-ALS_Part2-ALI- ENV– Airworthiness Limitation Section (ALS) Part 2 Airworthiness Limitation Items ALI Supplement Envelope (ENV), Rev05 or later revision	EASA 2023-0212	16/08/2024	Active
0085364		ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231 A320-214, A320-232		AD 2018-0091 is fully terminated by EASA STC 10071994 and equivalentlevel of safety as imposed by EASA AD 2018-0091 is achieved.If, during any inspection as defined in the P2F ALSPart 2 Supplement,any crack or damage in the AD affected area is found, before next flightcontact STC holder for approve		EASA STC 10071994 rev. 28EFW-E-AMOC-0027 Rev. A – Statementfor Alternative Method of ComplianceA320/ A321P2F-ALS_Part2-ALI- ENV – Airworthiness Limitation Section (ALS) Part 2 Airworthiness Limitation Items ALI Supplement Envelope (ENV), Rev08 or later revision.	EASA 2018-0091	24/09/2024	Active
0085365		ELBE FLUGZEUGWERK E GmbH	A321-211 AND A321-231		AD 2016-0106R2 is partly terminated by EASA STC 10071994. AD 2016-0106R2is fully terminated for the LHS for all A321P2Faircrafts and for theRHS for A321P2F aircrafts Pre-mod 155607 and the same level of safety asimposed by EASA AD 2016-0106R2 is achieved.lf, any crack or damage in the AD affected a		EASA STC 10071994 Rev. 28EFW-E-AMOC-0028 Rev. A – Statementfor Alternative Method of ComplianceA320/ A321P2F-ALS_Part2-ALI- ENV – Airworthiness Limitation Section (ALS) Part 2 Airworthiness Limitation Items ALI Supplement Envelope (ENV), Rev08 or later revision.	EASA 2016-0106R2	24/09/2024	Active
0085541		ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231 A320-214, A320-232		All requirements of AD 2020-0030, except the corrective actions perparagraph 2, are applicable to allA320/A321 A/Cs which have undergone the passenger to freighterconversion prescribed by EASASTC 10071994. All defined inspection thresholds(AD paragraph 1), credit(AD paragraph 3) and terminating act		EASA STC 10071994 rev. 28EFW-E-AMOC-00034 Rev. A – Statement for Alternative Method ofCompliance.	EASA 2020-0030	14/10/2024	Active



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10085655		ELBE FLUGZEUGWERK E GmbH	A321-231 A321-211 A320-232 A320-214	N/A	Time Limits / Maintenance Checks – Airworthiness Limitations SectionPart 3 – Certification Maintenance Requirements – AmendmentSuperseded EASA AD 2022-0091 required accomplishmentof all maintenancetasks as described in AIRBUS ALS Part 3 at Revision 08 Issue 02. EFWraised AMOC EFW-E- AMOC-00014 which		EASA STC 10071994 Rev. 28EFW-E-AMOC-00030 Rev. B – Statement for Alternative Method of ComplianceEFW Doc. A320/ A321P2FALS_PART3- CMR-ENV Supplement Rev 03	2024-0030	23/10/2024	Active
0085656		ELBE FLUGZEUGWERK E GmbH	A321-231 A321-211 A320-232 A320-214	N/A	Time Limits / Maintenance Checks – Airworthiness Limitations SectionPart 4 – Certification Maintenance Requirements – AmendmentSuperseded EASA AD 2022-0102 required accomplishmentof all maintenancetasks as described in AIRBUS ALS Part 4 at Revision 08 or at Revision 08Issue 02. EFW raised AMOC 349/		EASA STC 10071994 Rev. 28EFW-E-AMOC-00031 Rev. A – Statement for Alternative Method of ComplianceEFW Doc. A320/ A321P2FALS_PART4- CMR-ENV Supplement Rev 02	2024-0046	23/10/2024	Active
0085929		ELBE FLUGZEUGWERK E GmbH	A330-341/-342/-343 A330-321/-322/-323 A330-301/-302/-303 A330-223/-243 A330-201/-202/-203	The AMOC is valid in conjunction with the summary of investigationpresented with the EFW Document Ref. EFW-E- AMOC-00035dated 28.11.2024.The AMOC issue is based on the AD 2023-0199.	Time Limits / Maintenance Checks – Airworthiness Limitations SectionPart 3 – Certification Maintenance Requirements – AmendmentAirbus issued the ALS Part 3 Revision 08 to introduce new tasks andlimitations. For these reasons, EASA AD 2023-0199 requiresaccomplishment of the actions specified in the R		EFW Statement for Alternative Method of Compliance Ref. EFW-E- AMOC-00035Rev A dated 28.11.2024	2023-0199	29/11/2024	Active
0085993		ELBE FLUGZEUGWERK E GmbH	A321-211, A321-231		Alternative Method of Compliance to Airworthiness DirectiveEASA2023-0074All requirements of AD 2023-0074, except for the corrective actions perparagraph 3, remain applicable to all A321 A/Cs which have undergone thepassenger to freighter conversion prescribed by EASA STC approval10071994.If, during		EASA STC 10071994EFW- E-AMOC-00037 Rev.A – Statement for Alternative Method of Compliance.	2023-0074	09/12/2024	Active
0086155		ELBE FLUGZEUGWERK E GmbH	A321-231 A321-211 A320-232 A320-214	N/A	ATA 53 – Fuselage – Centre Fuselage Forward Pressure Bulkhead – InspectionAlternative Method of Compliance to Airworthiness Directive EASA2024-0147R1AD 2024-0147R1 is fully terminated by EASA STC 10071994 and equivalentlevel of safety as imposed by EASA AD 2024-0147R1 is achieved.lf, during any inspe		EASA STC 10071994EFW- E-AMOC-00033 Rev. B – Statement for Alternative Method of ComplianceA320/ A321P2F-ALS_PART2-ALI- ENVRev.08: A320-A321 P2F AIRWORTHINESSLIMITAT ION SECTION ALS PART 2 AIRWORTHINESS LIMITATION ITEMS ALISUPPLEMENT ENVELOPE (ENV)	2024-0147R1	09/01/2025	Active
0086156		ELBE FLUGZEUGWERK E GmbH	A321-231 A321-211	N/A	ATA 53 – Fuselage – Cabin Floor Beam Junction – InspectionAlternative Method of Compliance to Airworthiness Directive EASA2024-0128AD 2024-0128 is partly terminated by EASA STC 10071994.AD 2024-0128 is fully terminated for the LHS for all A321P2F aircraftdue to replacing of affected parts. For the R		EASA STC 10071994EFW- E-AMOC-00032 Rev.B – Statement for Alternative Method of ComplianceA320/ A321P2F-ALS_PART2-ALI- ENV Rev.08: A320-A321 P2F AIRWORTHINESSLIMITAT ION SECTION ALS PART 2 AIRWORTHINESS LIMITATION ITEMS ALISUPPLEMENT ENVELOPE (ENV)	2024-0128	09/01/2025	Active



EASA Certificate	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
Number 10086217		ELBE FLUGZEUGWERK E GmbH	A330-341/-342/-343 A330-321/-322/-323 A330-301/-302/-303 A330-223/-243 A330-201/-202/-203	N/A	ATA 05 - Time Limits / Maintenance Checks – Damage ToleranceAirworthiness Limitation Items – ALS Part 2 – AmendmentAirbus issued the ALS Part 2 Revision 06 Issue 2 to introduce new tasksand limitations. For these reasons, EASA AD 2024-0011requiresaccomplishment of the actions specified in the Revis		EASA STC 10063798EFW- E-AMOC-00039 Rev. A – Statement for Alternative Method of Compliancedated 16 December 2024A330P2F AIRWORTHINESS LIMITATION SECTION ALS PART 2 AIRWORTHINESSLIMITAT ION ITEMS ALI SUPPLEMENT ENVELOPE (ENV) Rev. 13	2024-0011	20/01/2025	Active
10086218		ELBE FLUGZEUGWERK E GmbH	A330-341/-342/-343 A330-321/-322/-323 A330-301/-302/-303 A330-223/-243 A330-201/-202/-203	N/A	Time Limits / Maintenance Checks – System Equipment MaintenanceRequirements – ALS Part 4 – AmendmentAirbus issued the ALS Part 4 Revision 09 to introduce new tasks andlimitations.For these reasons, EASA AD 2024-0014 requiresaccomplishmentof the actions specified in the Revision 09.Due to conversio		EASA STC 10063798EFW- E-AMOC-00036 Rev. A – Statement for Alternative Method of Compliancedated 13 November 2024A330P2F/ ALS_PART4-SEMR-ENV, A330P2F Aircraft Limitations Section Part 4 (SEMR) Supplement ENV Rev. 01.	2024-0014	20/01/2025	Active
10086219		ELBE FLUGZEUGWERK E GmbH	A321-231 A321-211 A320-232 A320-214	N/A	ATA 05 – Time Limits / Maintenance Checks – Airworthiness LimitationsSection – Part 2 – Damage Tolerant Airworthiness Limitation Items – AmendmentThe EASA AD 2024-0031 provides requirements for the accomplishment ofthe actions specified in the Airworthiness Limitations Section (ALS)AIRBUS A320FAM ALS		EASA STC 10071994EFW- E-AMOC-00038 Rev. A – Statement for Alternative Method of Compliancedated 16 December 2024EFW A320/ A321P2F-ALS_PART2-ALI- ENV, A320-A321P2F AirworthinessLimitation Section Part 2 (DT-ALI) Supplement ENV Rev.08.	2024-0031	20/01/2025	Active
10074281		EPI EUROPROP INTERNATIONAL GmbH	TP400-D6	None	Para (5) on page 3 of the referenced AD mandates that, from30 September2020, installation of an engine on an aeroplaneis allowed, provided theengine is a Group 2 engine, or has been modified in accordance with theinstructions of the related ASB. This AMOC revises the compliance timeof 31 March 202		Rolls-Royce Technical Report EDNS01000901392 issue 02.	EASA_AD_2019_0161-01	16/09/2020	Active
10077589		EPI EUROPROP INTERNATIONAL GmbH	TP400-D6	n/a	ATA 72 – Engine – Propeller Shaft – InspectionAlternative inspection method for FPI inspection of the PGB propellershaft.AD 2019-0013_C1 requires performing an FPI inspection of the PGBpropeller shaft (P/N TP600741) before reaching a life of 1984 ReferenceFlight Cycles. The manufacturer (GE AVIO) ha		RT-42M0200-E- EP-21-0021 issue 01	2019-0013-C1	17/11/2021	Active
10086680		EUROPEAN AIR TRANSPORT LEIPZIG GmbH	A300 B4-622R	All requirements and conditions of AD 2012-0044 that are notspecifically referenced above remain fully applicable and must becomplied with accordingly.This approval is applicable only to Aircraft Manufacturer Serial Number(MSN) 617.	AD 2021-0044 requires reinforcing the door frame shells of the passengerdoors 2 and 4 on both side of the fuselage in accordance with TC-holderService Bulletin (SB) in ref. a. of this AMOC associated technicaldocumentation.On door 4R in the Lower Forward area, aircraft D-AEAD (MSN 617) presenta rein		a. Airbus SB A300-53-6170 Rev. 01 Dated 17 Feb 2023, or later approvedrevisions.b. Airbus RDAF ref. 81401781/020/2024 Issue A.c. Airbus RDAF ref. 81401781/020/2024 Issue B.	2012-0044	13/03/2025	Active



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10086383		FRENCHBEE	TRENT XWB-79B, TRENT XWB-84 TRENT XWB-75, TRENT XWB-79	The operator has to adhere to the instructions of Rolls-Royce NMSB TRENTXWB 72-AK633 Revision 1 except point 3.A.(6) Acceptane Criteria.Instead, the Acceptance Criteria the operator is conditioned to are:(a) Follow Rolls-Royce NMSB TRENT XWB 72- AK633 Revision 1 point3.A.(6)(a).(b) If there are crack	EASA AD 2024-0167 references Rolls- Royce NMSB TRENT XWB 72- AK633Revision 1. Rolls-Royce has proposed a Revision 2 of the reference NMSBwhich has been concurred with by EASA. Rolls-Royce NMSB TRENT XWB72-AK633 Revision 2 has not been released. Revision 1 of the NMSBallowed only single time fly-on dep		EASA AD 2024-0167, dated 22 August 2024Rolls- Royce NMSB TRENT XWB 72-AK633 Revision 1Rolls- Royce TPCR SB:DB-9682 Issue 1, concurred by EASA 07 February 2025.	2024-0167	07/02/2025	Active
10075459		HELI-HOLLAND AIR SERVICE BV	EC 175 B	None	ATA 53 - Fuselage- Lower Fuselage Fuel Tank Structure and EquipmentElectrical Connectors - Rework/ RelocationAs Alternative Method of Compliance (AMOC) to the requirements of AD2020-0004, Paragraph (1):Before exceeding 880 flight hours or within 18 months after theeffective date of this AD, whicheve		AH NTO no. ETJ 21-06029, dated 22.01.2021	2020-0004	25/01/2021	Active
10072276	REV. 1	ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 with Serial Number (S/N) 27898.	Nose Wheel Well sidewall panel web (2016-06-07 The Boeing Company: Amendment 39-18438; Docket No.FAA-2014-0774; Directorate Identifier 2013-NM-154-AD)Israel Aerospace Industries has requested an AMOC to Nose Wheel Well(NWW) initial and repeat inspections mandated by Airworthiness Directive(AD) 2016-0	Model: 747-400; Serial No.: 27898	Israel Aerospace Industries crack repair, stress and DamageToleranceAnalysis (DTA) document CS-27898-010 Rev "A" datedApril 23, 2019	US 2016-06-07	13/08/2020	Active
10073466		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Escape Slides Latch ReplacementIsrael Aerospace Industries (IAI) has requested an AMOC to theinspections and rework of the latches in the main deck escape/ sliderafts and the escape slides installed on the airplane doors mandated byAirworthiness Directive (AD) 2013-19-13 affecting Boeing model 747-40			2013-19-13	10/06/2020	Active
10073467		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to theinspections for BMS 8-39 flexible urethane foam insulation in the floorpanel assemblies and the power drive unit (PDU) cover assembliesmandated by Airworthiness Directive (AD) 2018-16-06 affecting Boeingmodel 747-400 series airplane conve	747-400 - Aircraft modified to special freighter per EASA STC 10014423(modified from PAX) or per EASA STC 10015911 (modified from Combi).		EASA ADOPTED FAA AD 2018-16-06	10/06/2020	Active
10073474		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).Limited to the all cargo configuration only. Any passenger configurationremains subject to actions required by AD 2009-19-06	Flight deck security door improvementIsrael Aerospace Industries (IAI) has requested an AMOC to the flightdeck door modifications mandated by Airworthiness Directive (AD) 2019-09-06 affecting Boeing model 747-400 series airplane converted toIAI special freighter.IAI freighter conversion in accordance			2009-19-06	12/06/2020	Active



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0073475		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Hot short protector (HSP) for the fuel quantity indicating system (FQIS)Israel Aerospace Industries (IAI) has requestedan AMOC to theinstallation of a hot short protector (HSP) for the fuel quantityindicating system (FQIS) of the center fuel tank and, for certainairplanes, the horizontal stabilizer			AD 2010-20-12	12/06/2020	Active
0073489		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to theEnvironmental Control System (ECS) duct assembly rework or replacementinstructions mandated by Airworthiness Directive (AD) 2010-14-01affecting Boeing model 747-400 series airplane converted to IAI specialfreighter.During the IAI speciali		IAI removal instructions Report 366-21-00-A2407 issue New dated 22 Nov2005IAI removal instructions Report 366-21-00-A0076 Issue B dated 17 July2007.	EASA ADOPTED FAA AD 2010-14-01	15/06/2020	Active
0073492		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to themeasurement of the electrical bond resistance between themotor operatedvalve (MOV) actuators and airplane structure for the horizontalstabilizer fuel tank (HST) and applicable corrective actions mandated byparagraph (h) of AirworthinessB			EASA ADOPTED FAA AD 2011-06-03	15/06/2020	Active
0073497		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Wing Side of Body Joint, Horizontal Stabilizer Side of BodyJoint, andFuselage Circumferential Splice InspectionIsrael Aerospace Industries (IAI) has requested an AMOC to wing andhorizontal stabiliser side-of-body joints and fuselage circumferentialsplice initial and repeat inspections mandated by p		Israel Aerospace Industries, Inspection of 747-400 SF FuselageCircumferential Splices at IAI Straps, document 366-53-00- A5493 datedJune 23, 2014.	2010-05-03	16/06/2020	Active
0073513		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to theinstructions mandated by paragraph (g) and paragraph (h)(1) ofAirworthiness Directive (AD) 2015-11-05 affecting Boeing model 747-400series airplane converted to IAI special freighter.Paragraph (g) of AD 2015-11-05 requires removal of the		- IAI Service Bulletin 366-11-143 Rev 2 dated Nov. 2015 forAFT CargoCompartment Heating System Marker and Tape Installation IAI Service Bulletin 366-21-152 Rev 1 dated Jan. 2019 for AFT CargoCompartment Heating System ZTS Installation.	EASA ADOPTED FAA AD 2015-11-05	16/06/2020	Active
0073528		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX)	Israel Aerospace Industries (IAI) has requested an AMOC to theinspection for cracks at the main entry door 5 fuselage cutout at theupper aft corner as mandated by Airworthiness Directive (AD) 2013-19-15affecting Boeing model 747-400 seriesairplane converted to IAI specialfreighter.AD 2013-19-15 req		Israel Aerospace Industries 747-400SF report – SupplementalStructuralInsp ection Document TR 366-00-00-A0316 Issue N dated March 2020.	EASA ADOPTED FAA 2013-19-15	18/06/2020	Active
0073536		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Aircraft converted to special freighter by EASA STC 10014423 (modifiedfrom PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to fuselageinitial and repeat inspections mandated by Airworthiness Directive (AD)2010-01-01 affecting Boeing model 747-400series airplane converted toIAI special freighter.AD 2010-01-01 requires repetitive inspections for cracking of certainf			EASA ADOPTED FAA AD 2010-01-01	18/06/2020	Active



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10073832		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300/-400	The AMOC is only valid for the Boeing 737-300 and 737-400 airplanemodels modified in accordance with EASA STC 10016227 Rev. 5 or per EASASTC 10015542 Rev. 5.Supersedure of the EASAAD 2014-25-09 and 2015-06-07, may invalidatethis AMOC.	US-2014-25-09: Fuselage - Forward Airstair Stowage Doorway Skin Assemblyand Bear Strap – InspectionUS-2015-06-07: Forward Entry Doorway and Airstairs Cutout - InspectionThe EASA AD 2015-06-07, applicable to 737-100, 737-200, 737-200C,737-300, 737-400, and 737-500 series airplanes, requires inspectio		737-300 freighter conversion STC ST01566LA.737-400 freighter conversion STC ST01961SE.IAI SB 365-53-045 Revision 2, dated June 2020.IAI Report 365-53-00-B9658 Revision A dated January 23, 2019 - Implementation of Boeing SB 737-53-1058 and 737-53-1083 on 737-300 SFand 737-400 SF.	2014-25-09 / 2015-06-07	20/07/2020	Active
10073835		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	The AMOC is only valid for the Boeing 737-300 and 737-400 airplanemodels modified in accordance with EASA STC 10016227 Rev. 5 or per EASASTC 10015542 Rev. 5.Supersedure of the EASAEASA AD US-2008-17-13, may invalidate this AMOC.	The EASA AD US-2008-17-13, applicable to 737-100, 737-200, 737-200C, 737-300, 737-400, and 737-500 series airplanes, requires replacing theexisting straight- to-90-degree hose assembly for the Lavatory "A"water supply with a new straight hose assembly and a separate 90-degreeelbow fitting, in orderc		737-300 freighter conversion STC ST01566LA.737-400 freighter conversion STC ST01961SE.IAI SB 365-38-032, Revision 0, dated November 2011.	AD US 2008-17-13	20/07/2020	Active
10073836		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	The AMOC is only valid for the Boeing 737-300 and 737-400 airplanemodels modified in accordance with EASA STC 10016227 Rev. 5 or per EASASTC 10015542 Rev. 5.Supersedure of the EASA AD US-2015-02-26, may invalidate this AMOC.	Fuselage - Seat Track Link Assembly - Modification / Repair/ReplacementThe EASA AD US-2015-02-26, superseding AD 2013-24-13, applicable to737-100, 737-200, 737-200C, 737-300, 737-400, and 737-500 seriesairplanes, requires replacing the pivot link assembly, replacing theseat track link assemblies or		737-300 freighter conversion STC ST01566LA.737-400 freighter conversion STC ST01961SE.	US 2015-02-26	20/07/2020	Active
10073856		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	The AMOC is only valid for the Boeing 737-300 and 737-400 airplanemodels modified in accordance with EASA STC 10016227 Rev. 5 or per EASASTC 10015542 Rev. 5.Supersedure of the EASAAD US-2013-18-08, may invalidate this AMOC.	Fuselage - Fuselage - Forward Entry Door Skin Cutout - InspectionThe EASA AD US-2013-18-08, applicable to 737-200, 737-200C, 737-300,737-400, and 737-500 series airplanes, supersedes AD 2004- 18-06,retaining requirements for repetitive inspections to find fatiguecracking of certain upper and lower s		737-300 freighter conversion STC ST01566LA.737-400 freighter conversion STC ST01961SE.	US-2013-18-08	22/07/2020	Active
10073869		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	The AMOC is only valid for the Boeing 737-300 and 737-400 airplanemodels modified in accordance with EASA STC 10016227 Rev. 5 or per EASASTC 10015542 Rev. 5.Supersedure of the EASAAD US-2015-13-05 may invalidate this AMOC.	Cracking in the upper corners of the forward entry door skin cutoutThe EASA AD US-2015-13-05, applicable to 737-100, 737-200, 737-200C,737-300, 737-400, and 737-500 series airplanes, requires repetitiveinspections for cracking in the uppercorners of the forward entry doorskin cutout, and repair ifA		737-300 freighter conversion STC ST01566LA737-400 freighterconversion STC ST01961SE	US-2015-13-05	24/07/2020	Active
10073879		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	The AMOC is only valid for the Boeing 737-300 and 737-400 airplanemodels modified in accordance with EASA STC 10016227 Rev. 5 or per EASASTC 10015542 Rev. 5.Supersedure of the EASAEASA AD US-2016-16-13, may invalidate this AMOC.	Cracking in the fuselage skin along the chem-mill steps at stringersS-1R and S-2R, between STA 400 and STA 460The EASA AD US-2016-16-13, applicable to 737-100, 737-200, 737-200C,737-300, 737-400, and 737-500 series airplanes, requiring repetitiveinspections to detect cracks in the fuselage skin alon		737-300 freighter conversion STC ST01566LA.737-400 freighter conversion STC ST01961SE.IAI SB 365-38-032, Revision 0, dated November 2011.	US-2016-16-13	24/07/2020	Active





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10074143		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	-Only applicable to aircraft and areas affected by IAI STC EASA 10016227Rev. 5 or per IAI STC EASA 10015542 Rev. 5, as applicableAll other requirements of AD US-2017-22-04 except those stated aboveremain applicable. This AMOC does not constitute a terminating action toAD US- AD 2017-22-04 inspecti	Fuselage - Skin Panels - InspectionAD US-2017-22-04, applicable to 737-200, -200C, 300, -400, and -500series airplanes,requires inspection, repair, and replacement ofdis-bonded fuselage skin panels Boeing Service Bulletin 737-53A1349,original issue, dated August 23, 2016.During the IAI 737 conversi		-IAI Service Bulletin 365-53-050 "Boeing SB 737-53A1349 implementationon IAI Special Freighters", Rev 1 dated June 2020IAI DTA report TR 365-53-00-92485 – 737-300/-400 SF<(>,<) >Rev D.	US-2017-22-04	01/09/2020	Active
10074340		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to the continuedstructural integrity inspections mandated by Airworthiness Directive(AD) 2018-04-07 affecting Boeing model 747-400 series airplane converted to IAI special freighter.AD2018-04-07 requires revising the maintenance or inspection p		Israel Aerospace Industries 747-400SF report – SupplementalStructuralInsp ection Document TR 366-00-00-A0316 Issue N dated March 2020.	#MULTIVALUE	24/09/2020	Active
10074341		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to themodifications as per Boeing Service Bulletin (SB) 747-24A2193 Revision 1dated June 19 1997 mandated by paragraph (b) of Airworthiness Directive(AD) 99-06-18 affecting Boeing model 747-400 series airplane convertedto IAI special freighter		Boeing Service Bulletin (SB) 747-24A2193, Revision 1, datedJune 19,1997	FAA 99-06-18	24/09/2020	Active
10074407		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to the reworkand replacement requirements on the environmental control system (ECS)duct assemblies as per Boeing Alert Service Bulletin (SB) 747-21A2416,Rev 0, dated 29 June 2000 mandated by paragraph (a) of AirworthinessDirective (AD) 2000-26			FAA 2000-26-05	28/09/2020	Active
10074420		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to therepetitive inspections and repairs modifications as per Boeing ServiceBulletin (SB) 747-53A2416 Revision 1 dated May 6, 1999 mandated byAirworthiness Directive (AD) 2000-02-10 affecting Boeing model 747-400series airplane converted to IAI		Boeing Service Bulletin 747-53A2416, revision 3, dated April 25, 2002IAI Service Bulletin 366-53-121 Rev 0 dated September 2011	FAA 2000-02-10	29/09/2020	Active
10074421		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to therepetitive inspections and repairs modifications mandatedby paragraphs(e) and (g) of Airworthiness Directive (AD) 2001-11-06 affecting Boeingmodel 747-400 series airplane converted to IAI special freighter.AD 2001-11-06 requires repetiti		Boeing Service Bulletin 747-53A2390 Revision 1 dated July 6, 2000IAI Supplemental Structural Inspection Document (SSID)366-00-00- A0316,Rev. N, dated March 2020	FAA 2001-11-06	29/09/2020	Active
10074431		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to theinstallation and modifications instructions modificationsmandated byparagraph (f) of Airworthiness Directive (AD) 2009-15-12 affectingBoeing model 747-400 series airplane converted to IAI special freighter.AD 2009-15-12 requires installi			FAA 2009-15-12	30/09/2020	Active





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0074435		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to themodifications instructions of the inflation system of escape slides,ramp/slides and slides/raft mandated by Airworthiness Directive (AD) 2009-10-12 affecting Boeing model 747-400series airplane converted toIAI special freighter.AD 2009-10			FAA 2009-10-12	30/09/2020	Active
0074438		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to theinstructions of reconfiguring the clamps of certain wire bundles andapplying insulating sealant to certain fasteners inside the auxiliaryfuel tank, main fuel tanks and surge fuel tanks required by paragraph(f)(1) of Airworthiness Directiv			FAA 2007-20-01	01/10/2020	Active
0074439	REV. 1	ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10015911 (modified from Combi).	Israel Aerospace Industries (IAI) has requested an AMOC to themodification of the decompression panels on the smoke barrier in themain deck cargo compartment or replacement of thesmoke barrier with animproved smoke barrier and subsequent inspection requirements mandatedby paragraphs (f) and (g) ofa			FAA 2007-04-10	01/10/2020	Active
0074440		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modification from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to therequirements of replacing the cell stack of the flight deck humidifierwith a supplier tested cell stack or replacing the cell stack with ablanking plate and subsequently deactivating the flight deck humidifieras mandated by Airworthiness			FAA 2006-21-05	01/10/2020	Active
10074443		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to the revisionof the maintenance inspection program to includeinspections that willgive no less than the required damage tolerance rating for eachstructural significant item, as mandated by paragraphs (h) and (i) ofAirworthiness Directive (AD		IAI Supplemental Structural Inspection Document (SSID) 366-00-00- A0316,Rev. G, dated May 2012	FAA 2004-07-22 R1	01/10/2020	Active
0074456		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to thereplacement and inspection requirements for shear-pin restraints ofGoodrich evacuation slides mandated by Airworthiness Directive (AD) 2008-03-05 affecting Boeing model 747-400 series airplane converted toIAI special freighter.During the I			2008-03-05	02/10/2020	Active
0074466		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to thereplacement and inspection requirements for shear-pin restraints ofGoodrich evacuation slides mandated by Airworthiness Directive (AD) 2008-06-27 affecting the Goodrich evacuation slides fitted on Boeingmodel 747-400 series airplane conver			2008-06-27	05/10/2020	Active
0074479		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to thereplacement requirements for a certain flight deck door feature andrevision of the modification record placard mandated by AirworthinessDirective (AD) 2008-01-01 affecting the Boeing model 747-400 seriesairplane converted to IAI special f			2008-01-01	06/10/2020	Active





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10074518		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to theinstructions of inspections in the water and drain lines in the cargocompartments, in accordance with Boeing Service Bulletin (SB) 747-30A2080, Revision 3 dated June 14, 2007, asrequired byAirworthiness Directive (AD) 2007-19-16 affecting			2007-19-16	09/10/2020	Active
10074519		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to therepetitive inspections and repairs modifications as per Boeing ServiceBulletin (SB) 747-53A2688, revision 2, dated August 21, 2014 mandated byAirworthiness Directive (AD) 2016-07-31 affecting Boeing model 747-400series airplane convertedA		Boeing Service Bulletin 747-53A2688, revision 2, dated August 21, 2014IAI Service Bulletin 366-53-098 Rev 6 dated May 2020	2016-07-31	09/10/2020	Active
10074530		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Repetitive inspection and modification of the upper deck tension tiesSTA 1120 through 1220Israel Aerospace Industries (IAI) has requested an AMOC to theinspections, modifications and post- modification inspections of theupper deck tension ties and shear webs mandated by AirworthinessDirective (AD) 20		Boeing Alert Service Bulletin 747-53A2507, revision 1, dated January 14,2010IAI Service Bulletin 366-53-077 Rev 6 datedMarch 2013 - "FUSELAGE –Stretched Upper Deck Frame and Tension Tie - Inspection and Repair"IAI Service Bulletin 366-53-139 Rev 2 dated June 2016 - "Modification oftension ties and	EASA ADOPTED FAA AD 2016-05-12	12/10/2020	Active
10074531		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Replacement of tie rods of the center overhead stowage binsIsrael Aerospace Industries (IAI) has requested an AMOC to thereplacement requirements of specified tie rods of the center overheadstowage bins mandated by Airworthiness Directive (AD) 2006-13-09affecting the Boeing model 747-400 series airp		None	EASA ADOPTED FAA AD 2006-13-09	12/10/2020	Active
10074734		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Inspection for corrosion and cracks of the station 980 upper deck floorbeam. (2005-16-10 Boeing: Amendment 39-14215. Docket No. FAA-2005-21088;Directorate Identifier 2004-NM-267-AD).Israel Aerospace Industries (IAI) has requested an AMOC totheinspection for corrosion and cracks of the station 980 up		None	FAA 2005-16-10	03/11/2020	Active
10074735		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Control panels for the galley cart lift replacement or modification.(2005-10-19 Boeing: Amendment 39-14096. Docket No. FAA-2004-19532;Directorate Identifier 2004-NM-87-AD).Israel Aerospace Industries (IAI) has requested an AMOC to theinstructions for replacing or modifying the control panels for the		None	FAA 2005-10-19	03/11/2020	Active
10074747		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to theinvestigative and corrective actions, as applicable, for galley chillerboost fan wire bundle W4489, per Boeing Alert Service Bulletin (SB)747-21A2427, dated April 24, 2003 and as mandated by AirworthinessDirective (AD) 2004-07-20 affectin			2004-07-20	04/11/2020	Active



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10074748		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Discrepancies of the fuselage frame to tension tie joints at bodystations 1120 through 1220.(2005-05-08 Boeing: Amendment 39-13997; Docket No. FAA-2004-19470;Directorate Identifier 2003-NM-268-AD).Israel Aerospace Industries (IAI) has requested an AMOC to theinspections of the fuselage frame to tens		None	FAA 2005-05-08	04/11/2020	Active
10074762		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to therequirements of modification of the passenger entertainment system (PES)and revision of the Airplane Flight Manual (AFM) to ensure that theflight crew is able to remove electrical power from the entire PES whennecessary, as mandated by Ai			2001-14-15	05/11/2020	Active
10074763	REV. 1	ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to therequirements of inspection of the separation between the galley powerfeeder and static ground wiring, and the adjacent passenger oxygensystem tubing in the forward ceiling area above the door 4 galley, andcorrective actions where necessar			EASA ADOPTED FAA AD 98-06-29	12/01/2021	Active
10074770		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Longeron Extension Fitting Inspection, Repairs and Modification.(2020-15-14 The Boeing Company: Amendment 39-21177; Docket No.FAA-2020-0208; Product Identifier 2019-NM-209- AD).Israel Aerospace Industries (IAI) has requested an AMOC to theinstructions for inspections of the longeron extension fitting		Boeing Service Bulletin 747-53A2860 Revision 3 dated November 11, 2019IAI Supplemental Structural Inspection Document (SSID) 366-00-00-A0316,Rev. N, dated March 2020	US 2020-15-14	06/11/2020	Active
10075038		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	- All provisions of AD 2014-08-08 that are not specificallyreferencedabov e remain fully applicable and must be complied accordingly This AMOC is applicable to Boeing Model 737-300 series airplanesmodified in accordance with EASA STC 10015542 Rev.5 and to Boeing Model737-400 series airplanes modif	AD 2014-08-08 was issued to detect and correct fatigue Cracks in theskin at the lower corners of the forward entry doorway on airplanes thatdo not have an airstair door cutout.During the conversion to special freighter a new main deck cargo doorsurround structure is installed with an external double		IAI SB 365-53-042 Rev 1 dated July 2014FAA AMOC approval 12S-14-471	AD 2014-08-08	02/12/2020	Active
10075047		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	All provisions of AD 2015-16-08 that are not specifically referencedabove remain fully applicable and must be satisfiedaccordingly.This AMOC is applicable to Boeing Model 737-300series airplanesmodified in accordance with EASA STC 10015542 Rev.5 and to Boeing Model737-400 series airplanes modified	Crown Skin Lap Joint - Inspection for Lower Row Cracking atStringerS-4, Between BS 360 and BS 1016AD 2015-16-08 requires inspections of the lap joint at stringers S-4Rand S-4L along the entire length from body station (BS) 360 to BS 1016,in accordance with the Accomplishment Instructions of BoeingS		- STC EASA 10015542 Rev. 5- STC EASA 10016227 Rev. 5- FAA AMOC approval 120S-11-319	(FAA) AD 2015-16-08	03/12/2020	Active





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0075063		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	All provisions of AD 2020-17-04 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-300 airplanesmodified to a 737-300special freighter (SF) or 737-300 SF with rigid barrier (RB) configuration in accordance with EASA S	Equipment / Furnishings - Passenger Service Units and Lanyard Assemblieson the Life Vest Panels - InstallationAD 2020-17-04, applicable to 737-300, -400, and -500 series airplanes,requires installing lanyard assemblies on the passenger service units(PSU) and, for certain airplanes, on the life vestS		IAI report 365-25-00-92925 Rev FIAI report 365-25-00- B4716 Rev CFAA AMOC approval 790-20-12281	(FAA) AD 2020-17-04	03/12/2020	Active
10075094		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	- All provisions of AD 2016-11-04 that are not specificallyreferencedabov e remain fully applicable and must be satisfied accordingly This AMOC is applicable to Boeing Model 737-300 series airplanesmodified in accordance with EASA STC 10015542 Rev.5 and to Boeing Model737-400 series airplanes modi	AD 2016-11-04 requires repetitive inspections for cracking of the1.04-inch nominal diameter wire penetration hole, and applicable relatedinvestigative and corrective actions, in accordance with theAccomplishment Instructions of Boeing Alert Service Bulletin737-53A1279, Revision 2<(>,<)> dated AprilB		- IAI SB 365-53-036 Rev 2- FAA AMOC approval 120L-17-207	AD 2016-11-04	07/12/2020	Active
10075095		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	- All provisions of AD 2009-04-15 that are not specificallyreferencedabov e remain fully applicable and must be satisfied accordingly This AMOC is applicable to Boeing Model 737-300 series airplanesmodified in accordance with EASA STC 10015542 Rev.5 and to Boeing Model737-400 series airplanes modi	AD 2009-04-15, applicable to 737-300 and -400 series airplanes, requiresrepetitive internal eddy current and detailed inspections to detectcracked stringer tie clips between STA 360 to STA 907 and Stringer(S)-10L to S-10R; measuring the fastener spacing and the edge margin ifapplicable, and doing ap		- IAI SB 365-53-050 Rev 1- FAA AMOC approval 120S-14-440	2009-04-15	07/12/2020	Active
10075333		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX).	Israel Aerospace Industries (IAI) has requested an AMOC to theaccomplishment of the repetitive mid- and low-frequency eddy currentinspections for cracks in the overlapped skin panels in the fuselageskin lap joints in sections 41, 42, 44 and46, and corrective actions ifnecessary, for certain Boeing M			2006-22-09	06/01/2021	Active
10075352		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to theinspection of the attachment of the shoulder restraint harness to themounting bracket on certain observer and attendant seats to determine ifa C-clip is used in the attachment, and corrective action if necessary,in accordance with Boeingd			FAA 2006-26-13	08/01/2021	Active
10075378		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX).	Israel Aerospace Industries (IAI) has requested an AMOC to theaccomplishment of repetitive inspections for cracks in the overlapping (upper) skin of the upper fastener row of the lap joints of the fuselageskin in sections 41, 42, and 46; and related investigative andcorrective actions, if necessary,B		- Boeing Service Bulletin 747-53A2499, Revision 3, dated July 15, 2014- IAI Service Bulletin 366-53-118 dated April 2012	FAA 2016-04-02	12/01/2021	Active





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10075379		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX).	Israel Aerospace Industries (IAI) has requested an AMOC to theaccomplishment of the inspections for scribe lines in affected lap andbutt splices, wing-to-body fairing locations, external repair and cutoutreinforcement areas; and related investigative and corrective actions ifnecessary, in accordance			FAA 2012-04-09	12/01/2021	Active
10075560		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	- The AMOC is only valid for the Boeing 737-300 and 737-400airplanemodels modified in accordance with EASA STC 10016227 Rev. 5 or per EASASTC 10015542 Rev.5 All provisions of AD 2018-04-06 that are not specifically referencedabove remain fully applicable and must be satisfied accordingly.	The AD 2018-04-06, superseding AD 2012-12-05, requires repetitiveinspections of the intercostal webs, attachment clips,straps, stringersand stringer splice channels for cracks; and corrective action ifnecessary, at Body Stations BS 360 toBS 348.2 (AFT of the forward entrydoor) and BS 303.9 to BS 2		- IAI SB 365-53-020 revision 5, dated September 2020 FAA AMOC approval 790-20-14393. Dated 12 November 2020.	2018-04-06	02/02/2021	Active
0075595		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	All provisions of AD 2020-26-04 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-300 and 737-400 aeroplanes modified to aspecial freighter configuration in accordance with EASA STC 10015542 orEASA STC 10016227.	Fuselage - Skin Panels - Inspection / Repair / InstallationAD 2020-26-04, superseding AD 2013-18-08, requires repetitiveinspections for cracking of certain upper and lower skin panels of thefuselage and of the fuselage skin along certain chem-milled lines, andcorrective actions if necessary, in acco		FAA AMOC approval 120S-13-696	(FAA) 2020-26-04	09/02/2021	Active
0075596		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300	All provisions of AD 2005-03-02 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-300 airplanesmodified to a 737-300special freighter (SF) or 737-300 737-300 Quick Change (QC)configuration in accordance with EASA ST	Fuselage Open Body Station Frames - Inspection / Air ConditioningOverhead Ducts - ModificationAD 2005-03-02 requires inspection of the applicable body station frames,related investigative/ corrective actions, and installation of lanyardhookbrackets and lanyard assemblies under the air conditioning o		IAI Service Bulletin 365-21-007 Rev.0FAA AMOC approval 150S-07-274a	(FAA) 2005-03-02	09/02/2021	Active
0075597		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300	All provisions of AD 2006-26-09 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-300 airplanesmodified to a 737-300special freighter (SF) or 737-300 737-300 Quick Change (QC)configuration in accordance with EASA ST	Air Conditioning Outlet Extrusion Support Bracket - Inspection / Repair/ ModificationAD 2006-26-09 requires inspectionsfor cracks and modification repair ofcertain locations having Air Conditioning Brackets attachment inaccordance with Boeing SB 737-53-1216, Revision 1.The IAI 737-300 conversion fr		IAI Service Bulletin 365-53-008 Rev.3FAA AMOC approval 120L-17-191	(FAA) 2006-26-09	09/02/2021	Active





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Number 10075598		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300	All provisions of AD 2004-23-07 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-300 airplanesmodified to a 737-300special freighter configuration in accordance with EASA STC 10015542.	Number 2 Galley Fuselage Support Structure - ModificationAD2004-23-07 requires modification of certain fuselage supportstructure for the number2 galley, to prevent the galley from shifting, which could limit accessto the galley door during emergencies. During the IAI 737-300 conversionfrom passenge		Documentation FAA AMOC approval 100S-05-9	(FAA) 2004-23-07	lssue 09/02/2021	Active
10075599		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300	All provisions of AD 2003-26-08 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-300 airplanesmodified to a 737-300special freighter configuration in accordance with EASA STC 10015542.	Water / Waste - Potable Water Tank Pressure Relief Valve - ReplacementTo prevent rupture of the potable water tank during flight of theaeroplane, which could result in structural damage to the aeroplane andits inability to sustain flight loads, the AD 2003-26-08 requiresreplacing the existing pressu		FAA AMOC approval 100S-04-185	(FAA) 2003-26-08	09/02/2021	Active
10075625	REV. 1	ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	- All provisions of CAAI AD ISR-I-53-2019-12-6 R2 that are notspecifically referenced above remain fully applicable andmust becomplied with accordingly Only applicable to Boeing 737-300/-400 aeroplanes modified to specialfreighter per EASA STC 10016227 or per EASA STC 10015542.	CAAI AD ISR-I-53-2019-12-6 was prompted by a review of the manufacturingprocess for the 9G rigid barrier installed on IAI 737-300, 737-400passenger to freighter conversions, that identified a manufacturingnon-compliance.It has been found that the surface preparation before bonding might havebeen imp		IAI Service Bulletin 365-53-055 Rev. 0, dated August 2020.CAAI AMOC approval letter G333- D200820001, dated 20 August 2020.CS-27256-002 Rev. A, dated 10 February 2021.	CAAI AD ISR- I-53-2019-12-6 R2	22/02/2021	Active
10075655		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	- All provisions of AD 2012-26-02 that are not specificallyreferencedin this AMOC remain fully applicable and must be complied accordingly The AMOC is only valid for the Boeing737-300 and 737-400 airplanemodels modified in accordance with EASA STC 10016227 Rev. 5 or per EASASTC 10015542 Rev.5.	The AD 2012-26-02, supersedes AD 2005-13-27, retaining requirements forrepetitive inspections for cracking of the crownarea of the fuselageskin. This new AD modifies certain compliance times and also requiresrepetitive inspections for cracking using different inspection methods, inspecting additiona		EASA approved freighter conversions: - STC EASA 10016227 Rev. 5. - STC EASA 10015542 Rev. 5.	US AD 2012-26-02	17/02/2021	Active
10075701	REV. 1	ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300, 737-400	- All provisions of CAAI AD ISR-I-53-2019-12-6 R2 that are notspecifically referenced above remain fully applicable andmust becomplied with accordingly Only applicable to Boeing 737-300/-400 aeroplanes modified to specialfreighter per EASA STC 10016227.	CAAI AD ISR-I-53-2019-12-6 was prompted by a review of the manufacturingprocess for the 9G rigid barrier installed on IAI 737-300, 737-400passenger to freighter conversions, that identified a manufacturingnon-compliance.It has been found that the surface preparation before bonding might havebeen imp		IAI Service Bulletin 365-53-055 Rev. 0, dated August 2020.CAAI AMOC approval letter G333- D200820001, dated 20 August 2020.CS-28881-007 Rev -, dated 17 February 2021.CS-29206-005 Rev -, dated 17 February 2021.	(CAAI) ISR- I-53-2019-12-6 R2	22/02/2021	Active





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10075778		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300	All provisions of FAA AD 2004-03-34 R1 that are not specificallyreferenced above remain fully applicable and must be complied withaccordingly.Only applicable to Boeing 737-300 aeroplanes modified to specialfreighter per EASA STC 10015542, REV. 5.	Equipment / Furnishings - Door Mounted Escape Slide Latch CableAttachment Parts - ReplacementAD 2004-03-34 R1 requires replacing existing screw, nut, and washersthat attach the latch cable assembly to the latch block assembly of thedoor mounted escape slides, with new, improved screw, nut, and washe		Boeing Service Bulletin 737-25-1434, dated 22 March 2001.FAA AMOC approval letter 100S-04-185, dated 2 November 2004.	FAA 2004-03-34 R1	04/03/2021	Active
10075786		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300	All provisions of FAA AD 2002-07-10 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-300 aeroplanes modified to specialfreighter per EASA STC 10015542, REV. 5.	AD 2002-07-10 requires replacement of certain repairs in certainfuselage lap joints with improved repairs.During the IAI 737-300 Special Freighter conversion, a section of thefuselage is removed from BS 328 toBS 500B, between stringers 4R and 21L and is replaced by a new doorsurround structure. Thej		IAI Engineering Order El 365-53-00-99604 Issue New, dated 1November2004.FAA AMOC approval letter 100S-05-9 dated 26 January 2005.	(FAA) 2002-07-10	05/03/2021	Active
10075787		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-300	All provisions of FAA AD 2003-14-06 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-300 aeroplanes modified to specialfreighter per EASA STC 10015542, REV. 5.	AD 2003-14-06 requires repetitive inspections for cracking of certainlap splices, and corrective action if necessary.During the IAI 737-300 Special Freighter conversion, a sectionof thefuselage is removed from BS 328 toBS 500B, between stringers 4R and 21L and is replaced by a new doorsurround stru		IAI Engineering Order El 365-53-00-93774 Issue B, dated 27 October 2004.FAA AMOC approval letter 100S-05-9 dated 26 January 2005.	(FAA) 2003-14-06	05/03/2021	Active
10075892		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-400	All provisions of CAAI AD ISR-I-53-2019-12-6 R2 that are notspecifically referenced above remain fully applicable and must becomplied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified to specialfreighter per EASA STC 10016227.	CAAI AD ISR-I-53-2019-12-6 was prompted by a review of the manufacturingprocess for the 9G rigid barrier installed on IAI 737-400 passenger tofreighter conversion, that identifieda manufacturing non- compliance.It has been found that the surface preparation before bonding might havebeen improperly d		IAI Service Bulletin 365-53-055 Rev.0, dated August 2020.CAAI AMOC approval letter G333- D200820001, dated 20 August 2020.CS-27131-008 Rev -, dated 15 March 2021.	(CAAI) ISR- I-53-2019-12-6 R2	18/03/2021	Active
10076220		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-400	- All provisions of CAAI AD ISR-I-53-2019-12-6 R2 that are notspecifically referenced above remain fully applicable andmust becomplied with accordingly Only applicable to Boeing 737-400 aeroplanes modified to specialfreighter per EASA STC 10016227.	CAAI AD ISR-I-53-2019-12-6 was prompted by a review of the manufacturingprocess for the 9G rigid barrier installed on IAI 737-400 passenger tofreighter conversion, that identifieda manufacturing non- compliance. It has been found that the surface preparation before bonding might havebeen improperly d		IAI Service Bulletin 365-53-055 Rev.0, dated August 2020.CAAI AMOC approval letter G333- D200820001, dated 20 August 2020.CS-29205-004 Rev -, dated 6 April 2021.	(CAAI) ISR- I-53-2019-12-6 R2	14/04/2021	Active
10076732		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	747-400	Boeing Model 747-400 converted to IAI special freighter perEASA STC10014423 (modified from PAX) or per EASA STC 10015911 (modified fromCombi).	Israel Aerospace Industries (IAI) has requested an AMOC to therepetitive inspections of trim air diffuser ducts or sidewall riser ductassemblies (collectively referred to as TADDs) for damage and applicableon-condition actions, per Boeing Alert Service Bulletin (SB) 747-21A2577, Rev B dated FebruaryA			EASA ADOPTED FAA AD 2021-07-09	22/06/2021	Active





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10076920		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-400 737-300	All provisions of CAAI AD ISR-I-53-2019-12-6 R2 that are notspecifically referenced above remain fully applicable and must becomplied with accordingly.Only applicable to Boeing 737-300/-400 aeroplanes modified to specialfreighter per EASA STC 10015542 or per EASA STC 10016227.	Loading restrictions due to 9G barrier manufacture non-complianceCAAI AD ISR-I-53-2019-12-6 was prompted by a review ofthe manufacturingprocess for the 9G rigid barrier installedon IAI 737-300/-400 passengerto freighter conversions, thatidentified a manufacturingnon-compliance.It has been found t		IAI Service Bulletin 365-53-055 Rev. 1, dated April 2021.CAAI AMOC approval letter G333- D030521001, dated 3 May 2021.	ISR-I-53-2019-12-6 R2	13/07/2021	Active
10083154		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-400	All provisions of FAA AD 2013-09-01 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to aircraft modified tospecial freighter per STC EASA10016227 Rev. 5.	Fuselage - Lap Joints and Stringer Clips - Inspection / RepairFAA AD 2013-09-01 mandates the SB 737-53A1255 which requires inspections in the area between BS 328 and BS 500C extending from the S-21 L lapjoint upward through the S-4R lap joint inclusive. These inspectionswere previously included in th		-STC EASA 10016227 Rev. 5Boeing Alert Service Bulletin 737-53A1255, Revision 2-Boeing Alert Service Bulletin SB 737-53A1177-FAA AMOC 120S-10-162-EASA-TCDS- AS.02479_(IM) _IAI_B737400SF-02-1605 2012	FAA 2013-09-01	31/10/2023	Active
10084434		ISRAEL AEROSPACE INDUSTRIES, LTD.IAI	737-400	Applicable to IAI converted 737-400SF, S/N 23865	AMOC to AD 2017-04-08 as per EASA repair approval 10053132 rev 1.	The Type and OSD Certification Bases (CB) for the original productremains applicable to this certificate/ approval.The requirements for environmental protection and the associatedcertified noise and/ or emissions levels of the product are unchangedand remain applicable to this certificate/ approval w	IAI Repair Instruction CS-23865-004 Rev Bor later revisionsof the above listed document(s) approved/acceptedunder the EASA system.	2017-04-08	07/05/2024	Active
10075992		JET AVIATION AG	A319-115	None	Deviations from SB No. A320-53-1262 installation instructions perJBSL-M-03629 applicable in the locations modified by JBSL duringInterior Completion of the aircrafts (ref. FOCA STC 25-20-99 / LBA EMZTA0740 & EASA-IM.A.S.01185) Deviations are:- adaptation of the JBSL modified partition instead of the		- AIRBUS SB A320-53-1262- JBSL- M-03629	2016-0212	30/03/2021	Active
10073562	REV. 1	LEONARDO S.p.A.	AW 189 AND AW 169	This AMOC is valid for the helicopters AW169 serial numbers:69007, 69043, 69044, 69059, 69062, 69069, 69070, 69089, 69094 and 69096and for helicopters AW189 serial numbers:49007, 49008, 49011, 89001, 89002 and from 92001 to 92010.	Note: Rev. 1 cancels and supersedes the initial issue of thiscertificate. Repetitive Thermal Strip Installation: As alternative means of compliance to the requirement of § (3) of the EASA AD 2020-0048, accomplish the following action: (3) For Group 1 and Group 2 helicopters: Within 20 FH after lastacc		Annex 1 - Modification to the Leonardo Leonardo S.p.A. ASB 189-237original issue dated 29 May 2019, Revision A dated 05September 2019,and Revision B dated 04 February 2020 or Revision B dated04 February 2020 with Errata Corrige.Annex 2 - Modification to the Leonardo S.p.A. ASB 169-148 original issu	2020-0048	23/06/2020	Active
10086404		LEONARDO S.p.A.	AW189		As Alternative Method of Compliance (AMOC) to the requirement ofParagraph (1) of EASA AD 2024-0024, accomplish the following action:(1) For Group 1 aircraft (as defined in AD 2024-0024): Within 30 monthsafter the effective date of EASA AD 2024-0024, remove each affected part(as defined in AD 2024-00		Technical justification as reported in APPL/2025-013 dated 07/02/2025.	2024-0024	11/02/2025	Active



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0086405		LEONARDO S.p.A.	AW109SP		As Alternative Method of Compliance (AMOC) to the requirement ofParagraph (1) of EASA AD 2024-0024, accomplish the following action:(1) For Group 1 aircraft (as defined in AD 2024-0024): Within 30 monthsafter the effective date of EASA AD 2024-0024, remove each affected part(as defined in AD 2024-00		Technical justification as reported in APPL/2025-014 dated 07/02/2025.	AD 2024-0024	11/02/2025	Active
0073827		LUFTHANSA CARGO AG	777F	None	Fuel - Center Wing Tank Fuel Quantity Indicating System - Check / AFMRevisionLUFTHANSA CARGO AG submitted an application to EASA for an AMOC to ADUS-2020-11-11.AD US-2020-11-11 paragraphs (g)(1) through (6) define work instructionsto be accomplished by performing a Refueling Station Door Cycling (RS		Flight Crew Operations Manual Bulletin for Lufthansa Cargo AG No. LUB-32R1 date June 17, 2020Maintenance TIP No. 777 MT28-031 R2 dated May 29, 2020	US-2020-11-11	17/07/2020	Active
0080738		LUFTHANSA TECHNIK AG	A320-212 A319-113, A319-114, A320-211 A319-113/-114/-211/-212		AD 2018-0212R1 assigns a life limit to the forward engine mount titaniumcrossbeams p/n 238-0204-501 and requires the accomplishment of theAirbus Service Bulletin (SB) A320-71-1073for their replacement.Alternatively, an enhanced design (p/n 238-0204-503) is proposed andsupported by the accomplishmen		- AIRBUS SB A320-71-1076- GOODRICH AEROSTRUCTURES SB RA32071-171- LHT Jobcard TZ244404-77.	#MULTIVALUE	25/11/2022	Active
0086316		LUFTHANSA TECHNIK AG	737-800	All provisions of AD 2009-12-06 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-800 airplane,s/n 32971.	The EASA adopted FAA AD 2009-12-06 was issued to ensure that the flightcrew is able to turn off electrical power to IFEsystem and othernon-essential electrical systems through a switch in the flightcompartment.The engineering bulletin 737-EB24-0126 R2 has been implemented on737-800 MSN 32971 as par		Lufthansa Technik Engineering Bulletin 737- EB24-0126 R2FAA AMOC approval letter 783-22-6231, dated 18 April 2022.	(FAA) AD 2009-12-06	29/01/2025	Active
0076146		MHI RJ AVIATION ULC	ALL	This EASA AMOC is only valid for the CL-600 Regional Jet Series aircraftregistered in European Union Member States or in EASA associatedcountries (Iceland, Liechtenstein, Norway and Switzerland) and equippedwith Rockwell Collins Pro Line 4Flight Management System FMC-4200Rockwell Collins Part Numbe	Flight Management Computing Hardware System / Airplane Flight Manual - RevisionThe Airworthiness Directive 2020-10-05 was issued on May 25, 2020 by theFAA and endorsed by EASA.The component AD 2020-10-05 was prompted by reports of the FlightManagement Computer (FMC) software issuing incorrect turn co		MHI RJ Aviation ULC's AFM content as follows:- Airplane Flight Manual (AFM) CSP-B-012 for aircraft model CL-600-2C10 -Chapter 2, LIMITATIONS - Navigation Systems Limitations- prohibition ofuse of the FMS Temperature Compensation feature - first introducedthrough AFM Revision Submission RS-329, datedi	US-2020-10-05	09/04/2021	Active
0076749		MHI RJ AVIATION ULC	CL-600-2E25 (RJ SERIES 1000) CL-600-2D24 (RJ SERIES 900) CL-600-2D15 (RJ SERIES 705) CL-600-2C10 (RJ 700, 701, 702)	This EASA AMOC is only valid for the CL-600 Regional Jet Series aircraftserial number listed in the MHI RJ Aviation Service Bulletin670BA-34-053 and registered in European Union Member States or in EASAassociated countries (Iceland, Liechtenstein, Norway and Switzerland) and equipped with Rockwell Co	Alternative Method of Compliance to Airworthiness DirectiveUS-2020-10-05.The Airworthiness Directive 2020-10-05 was issued on May 25th, 2020 bythe FAA and endorsed by EASA.This component AD 2020-10-05 was prompted by reports of the FlightManagement Computer (FMC) software issuing incorrect turn comm		MHI RJ Aviation Service Bulletin 670BA-34-053 "Navigation -FlightManagement Computer - Removal of the Flight Management System (FMS)Temperature Compensation Function" Initial Issue, dated June 18th 2021or later approved revisions.The content of MHI RJ Service Bulletin 670BA-34-053 Initial Issue isb	FAA AD 2020-10-05	23/06/2021	Active





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10084827		NEOS S.p.a.	TRENT 1000-R3 TRENT 1000-P3, TRENT 1000-Q3 TRENT 1000-M3, TRENT 1000-N3 TRENT 1000-K3, TRENT 1000-L3 TRENT 1000-H3, TRENT 1000-J3 TRENT 1000-D3, TRENT 1000-G3 TRENT 1000-AE3, TRENT 1000-CE3	This AMOC can only be used for engines that have already accomplished atleast one inspection per NMSB 72-AK316 Revision4 or for engines thathave accumulated less than 600 flight cycles.	ATA 72 – Engine – High Pressure Turbine Blades – Inspection/ De- PairingLimitationEASA AD 2019-0099R2 references NMSB 72-AK316. Since the AD was issuedthe NMSB was revised to revision 4 and effectively AD 2019-0099R2 wassuperseded by AD 2024-0188. This AMOC allows the requirementsutilization of EASA		EASA AD 2019-0099R2, dated September 06th 2023EASA AD 2024-0118, dated June 25th, 2024Trent 1000 NMSB 72-AK316 Rev 04 dated May 10th, 2024	2019-0099R2	28/06/2024	Active
10075384	REV. 1	NOORDZEE HELIKOPTERS VLAANDEREN N.	AS 365 N3	Limited to helicopters equipped with a Goodrich hoist having a PartNumber (P/N) as listed in Table 1 of AD2015-0226R5 and installing a slip clutch MP/ N 44314-398.	As Alternative Method of Compliance (AMOC) to the requirements of AD2015-0226R5, Paragraph (7), Table 4, Interval applicable to All AffectedHelicopters, apply the following action:At intervals of 36 months (+ 4 months tolerance), or 2000 hoistoperating cycles, perform a mandatory check (CHK) of theM		EASA approval no. 10074085.AH ASB AS365-25.01 .65, Revision4.	2015-0226 R05	13/01/2021	Active
10075130		PARC AVIATION ENGINEERING SERVICESLIMITE D	A320-232	Applicable to A320-232 S/ N 4630.	Background:EASA AD 2015-0051 was prompted by reports of cracks affecting the aftpylon moveable fairing Rib 5. A regime of repetitive inspections wasrequired by this AD, until the embodiment of an optional modificationsupported by the Airbus SB A320-54-1028 accomplishment.The aft pylon moveable fairi		Airbus SB A320-54-1028 Revision 1, or later approved revision.IndiGo Working Orders ref : 7028513, dated 08 April 2016.	2015-0051	10/12/2020	Active
10075131		PARC AVIATION ENGINEERING SERVICESLIMITE D	A320-232	Applicable to A320-232 S/ N 4757.	Background:EASA AD 2015-0051 was prompted by reports of cracks affecting the aftpylon moveable fairing Rib 5. A regime of repetitive inspections wasrequired by this AD, until the embodiment of an optional modificationsupported by the Airbus SB A320-54-1028 accomplishment.The aft pylon moveable fairi		Airbus SB A320-54-1028 Revision 1, or later approved revision.IndiGo Working Orders ref : 7028517, dated 04 November 2018.	2015-0051	10/12/2020	Active
10079208		PARC AVIATION ENGINEERING SERVICEST/A CAE PARC AVIATION ENGINEERINGSE RVICES	A320-232	Limited to Serial Number: MSN 6311	ATA 25 – Equipment / Furnishings – Galley / Container End Stop – ReplacementThe AD 2022-0026 mandates the installation of door securing mechanismson some Safran / Zodiac Galleys G5P/Ns, through the accomplishment ofthe Airbus SB A320-25-1BF6, revision 01 and the Zodiac Galley EuropeService BulletinT		Airbus SB A320-25-1BF6< (>,<)> revision 01Safran Service Bulletin 213510-25-001, revision DZodiac Galley Europe Service Bulletin 601858-25-005001-002.	2022-0026	09/05/2022	Active
10083902		PIAGGIO AVIATION S.P.A.	AVANTI II	All provisions of AD 202-0122 that are not specifically referenced aboveremain fully applicable and must be complied with accordingly.Only applicable to Piaggio P180 Avanti II MSN1002.	ATA 55 – Stabilizers – Vertical Stabilizer Assembly – InspectionAD 2023-0122 requires inspections for stress corrosion at the fittingassembly of the horizontal tail trim actuator (HTTA). The AD alsorequires corrective actions depending on the case of corrosion detected.On MSN 1002, which is an aerop		DMT RP80-0421 rev. 0 MSN 1002 HTTA fitting crack – Temporary Repair	2023-0122	14/02/2024	Active



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10078180		PIPER AIRCRAFT, INC.	PA-32RT-300T (TURBO LANCE II) PA-32RT-300 (LANCE II) PA-32R-300 (LANCE) PA-32-300 (CHEROKEE SIX 300) PA-32-260 (CHEROKEE SIX 260) PA-28RT-201T (TURBO ARROW IV) PA-28RT-201 (ARROW IV) PA-28R-201T (TURBO ARROW III) PA-28R-201 (ARROW III) PA-28R-200 (ARROW II) PA-28R-200 (ARROW II) PA-28R-200 (ARROW) PA-28-180 (ARROW) PA-28-180 (ARROW) PA-28-181 (ARCHER III) PA-28-181 (ARCHER III) PA-28-161 (WARRIOR III) PA-28-161 (WARRIOR III) PA-28-161 (CADET) PA-28-151 (CHEROKEE WARRIOR)		AMOC to AD 2021-0107R1 to allow Piper Service Bulletin 1345revision A(SB 1345A) as alternative to Service Bulletin 1345 (SB 1345). SB 1345Aincorporates changes to expand the inspection procedures to improve thequality of the data obtained. Instructions are added to remove andorient the wing for bet		Service Bulletin 1345A.	2021-0107R1	21/01/2022	Active
10080998		PIPER AIRCRAFT, INC.	PA-32R-300 (LANCE)	Applicable to PA-32R-300 (Lance), S/N 32R-7680107.	AMOC to AD 2021-0107R2 to allow repair the mechanical damage present incertain bolt holes of the wing spar by oversizing the affected holes and installing larger bolts than required by paragraph 7 of AD 2021-0107R2, according to the instructions in the following engineering order:Piper Repair Engineer		Piper Repair Engineering Order PA-32R-300 (7680107)/01 Rev.–, dated8/12/2022.	2021-0107R2	06/01/2023	Active
10073287		PRECISION CONVERSIONS, LLC	757-200	Applicable to 757-200 MSN 25297 only.This AMOC is valid for12 months from 17/06/2019 onward. Continuedvalidity after the expiration date is conditioned to the approval of therelated Damage Tolerance data, and to any change in the ICAs, ifany,resulting from this approval being incorporated in the o	Fuselage - Crown Skin Panel - Inspection / Repair / ReplacementAMOC to AD US-2017-20-05 for MSN 25297 - stage 1Precision Engineering, LLC Engineering Authorisation ref. EA-19-1031Rev. IR, dated 06/01/19, documenting the replacement on certainB757-200, previously converted from "Passenger" to "Freigh		A-Precision Engineering EA-19-1031, Rev. IR, dated 06/01/19B-Precision Engineering EA-19-1031- SSR, Rev. IR, dated 06/01/19	US-2017-20-05	18/05/2020	Active
10073288		PRECISION CONVERSIONS, LLC	757-200	Applicable to 757-200 MSN 24617 only.This AMOC is valid for12 months from 07/06/2019 onward. Continuedvalidity after the expiration date is conditioned to the approval of therelated Damage Tolerance data, and to any change in the ICAs, ifany,resulting from this approval being incorporated in the o	Fuselage - Crown Skin Panel - Inspection / Repair / ReplacementAMOC to AD US-2017-20-05 for MSN 24617 – stage 1Precision Engineering, LLC Engineering Authorisation ref. EA-19-1025Rev. C, dated 05/23/19, documenting the replacement on certain B757-200,previously converted from "Passenger" to "Freight		A-Precision Engineering EA-19-1025, Rev. C, dated 05/23/19B-Precision Engineering EA-19-1025- SSR, Rev. C, dated 05/23/19	US-2017-20-05	18/05/2020	Active



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10073289		PRECISION CONVERSIONS, LLC	757-200	Applicable to 757-200 MSN 25297 only.This AMOC is conditioned to the approval of the related Damage Tolerancedata, and to any change in the ICAs, if any, resulting from thisapproval being incorporated in the operator's maintenance manuals.	Fuselage - Crown Skin Panel - Inspection / Repair / ReplacementAMOC to AD US-2017-20-05 for MSN 25297 - stage 2 and 3Precision Engineering, LLC Engineering Authorisation ref. EA-19-1031Rev. IR, dated 06/01/19, documenting the replacement on certainB757-200, previously converted from "Passenger" to "		A-Precision Engineering EA-19-1031, Rev. IR, dated 06/01/19B-Precision Engineering EA-19-1031- SSR, Rev. IR, dated 06/01/19C-Precision Engineering EA-19-1031- DTE, Rev. IR, dated 07/26/19	US-2017-20-05	18/05/2020	Active
10073290		PRECISION CONVERSIONS, LLC	757-200	Applicable to 757-200 MSN 24617 only.This AMOC is conditioned to the approval of the related Damage Tolerancedata, and to any change in the ICAs, if any, resulting from thisapproval being incorporated in the operator's maintenance manuals.	Fuselage - Crown Skin Panel - Inspection / Repair / ReplacementAMOC to AD US-2017-20-05 for MSN 24617 – stage 2 and 3.Precision Engineering, LLC Engineering Authorisation ref. EA-19-1025Rev. C, dated 05/23/19, documenting the replacement on certain B757-200,previously converted from "Passenger" to "		A-Precision Engineering EA-19-1025, Rev. C, dated 05/23/19B-Precision Engineering EA-19-1025- SSR, Rev. C, dated 05/23/19C-Precision Engineering EA-19-1025- DTE, Rev. IR, dated 07/25/19	US-2017-20-05	18/05/2020	Active
10073291		PRECISION CONVERSIONS, LLC	757-200	Applicable to 757-200 MSN 25294 only.This AMOC is conditioned to the approval of the related Damage Tolerancedata, and to any change in the ICAs resulting from this approval beingincorporated in the operator's maintenance manuals. These data includethe skin replacement, if any, and those resulting f	Fuselage - Crown Skin Panel - Inspection / Repair / ReplacementAMOC to AD US-2017-20-05 for MSN 25294.Precision Engineering, LLC Engineering Authorisations ref. EA-19-1030,Rev. IR, dated 06/01/19 & EA-19-1012, Rev. IR, dated 03/07/19,documenting the replacement on B757-200 MSN 25294, previously conv		A-Precision Engineering EA-19-1030, Rev. IR, dated 06/01/19B-Precision Engineering EA-19-1030- SSR, Rev. IR, dated 06/01/19C-Precision Engineering EA-19-1030- DTE, Rev. IR, dated 07/26/19D-Precision Engineering EA-19-1012, Rev. IR, dated 03/07/19E- Precision Engineering EA-19-1012-SSR, Rev. IR, dated 0	US-2017-20-05	18/05/2020	Active
10073292		PRECISION CONVERSIONS, LLC	757-200	Applicable to 757-200 MSN 27051 only.This AMOC is conditioned to the approval of the related Damage Tolerancedata , andto any change in the ICAs, if any, resulting from thisapproval being incorporated in the operator's maintenance manuals.	Fuselage - Crown Skin Panel - Inspection / Repair / ReplacementAMOC to AD US-2017-20-05 for MSN 27051.Precision Engineering, LLC Engineering Authorisation ref. EA-19-1032,Rev. IR,dated 07/01/19, documenting the replacement on certainB757-200, previously converted from "Passenger" to "Freighter" per		A-Precision Engineering EA-19-1032, Rev. IR, dated 07/01/19B-Precision Engineering EA-19-1032- SSR, Rev. IR, dated 05/23/19C-Precision Engineering EA-19-1032- DTE, Rev. IR, dated 07/26/19	US-2017-20-05	18/05/2020	Active
10073406		PRECISION CONVERSIONS, LLC	757-200	Only applicable to 757-200 converted to freighter configuration perPrecision Conversions STC ST01529SE (EASA STC 10015539).Completion of this service bulletin as an AMOC only addresses aircraft(or individual side (s) of an aircraft) for which no crack is foundduring the inspections. For on- conditionP	Fuselage - Fuselage Frame Web - InspectionAirworthiness Directive US-2019-23-06 is applicable to Boeing 757-200,-200CB,and -300 series airplanes, certificated in any category. The ADrequires open-hole HFEC inspections at STA 1640 for any crack of thefuselage frame web fastener holes, on the left an		Ref. A - Precision Conversions, LLC SERVICE BULLETIN PC-757-53-A0112<(>,<)> "Supplemental Instructions for Inspection of Station 1640 Fuselage Frame Web at S-14", Rev. 1, dated May 29,2020	US-2019-23-06	04/06/2020	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
0073478	REV. 1	PRECISION CONVERSIONS, LLC	757-200	Applicable to B757-200 MSN 27051 only.This AMOC is valid for 12 months from 12 May 2020 onward. Continuedvalidity beyond this initial 12 months period is conditioned to theapproval of the related repair Fatigue & Damage Tolerance data<(>,<)>and to any change in the ICAs resulting from this approvali	Fuselage – Fuselage Frame Web STA 1640 - InspectionFuselageFrame Web STA 1640 – MSN 27051 – stage 1MSN 27051 is converted from a passenger configuration to a cargoconfiguration by Precision Conversions STC ST01529SE (EASA STC10015539). Cracks were found common to S-14R OEM intercostal clipfasteners		A-Precision Engineering EA-20-1007, Rev. IR, dated 05/12/20B-Precision Engineering EA-20-1007- SSR, Rev. IR, dated 05/12/20	(AD) US-2019-23-06	16/06/2020	Active
0073544		PRECISION CONVERSIONS, LLC	757-200	-Only applicable to MSN 24614-Aircraft converted to freighter configuration per Precision ConversionsSTC ST01529SE (EASA STC 10015539)	Time Limits / Maintenance Checks - Airworthiness LimitationsInstructions - Principal Structural Elements - Inspections / Maintenanceor Inspection Program RevisionAirworthiness Limitations Instructions - Principal Structural Elements -InspectionsThe subject aircraft is converted from a passenger conf		A - Precision Engineering document EA 19-1038 Rev A, B757-200 – ExistingExternal Repair at STA 350 and S-4LB - Precision Engineering document EA 19-1038-SSR Rev A, B757-200 –Existing External Repair at STA 350 and S-4LC - Precision Engineering document EA 19-1038-DTE Rev. IR, B757-200 –Existing Exte	US-2020-01-18	19/06/2020	Active
0074050		PRECISION CONVERSIONS, LLC	757-200	Applicable to 757-200 MSN 25056 only.	Fuselage - Crown Skin Panel - Inspection / Repair / ReplacementAMOC to AD US-2017-20-05 for MSN 27056MSN 27056 is converted from passenger to freighter by the PrecisionConversionsSTC ST01529SE (EASA STC 10015539) and has been modified bythe lessor as listed in AD US-2017-20-05<(>,<)> paragraph (k)(A-Precision Engineering EA 20-1006, Rev. IR, dated 06/22/2020B-Precision Engineering EA 20-1006- SSR, Rev. IR, dated 06/22/2020C-Precision Engineering EA 20-1006- DTE, Rev. IR, dated 07/08/2020D- Precision document ENG180204 Revision 2<(>,<)> Customer ServiceLetter (No Technical Objection)	US 2017-20-05	17/08/2020	Active
0074052		PRECISION CONVERSIONS, LLC	757-200	Applicable to B757-200 MSN 27054 only.This AMOC is valid for 12 months. Continued validity after theexpiration date is conditioned to the approval of the related DamageTolerance data, and to any change in the ICAs, if any<(>,<)> resultingfrom this approval being incorporated in the operator's mainte	Fuselage - Crown Skin Panel - Inspection / Repair / ReplacementAMOC to AD US-2017-20-05 for MSN 27054 – stage 1MSN 27054 is converted from passenger to freighter by the PrecisionConversions STC ST01529SE (EASA STC 10015539) and has been modified bythe lessor as listed in AD US-2017-20-05, paragraphT		A-Precision Engineering EA 20-1028, Rev. IR, dated 6/22/2020B-Precision Engineering EA 20-1028 SSR, Rev. IR, dated 6/22/2020C- Precision document ENG180204 Revision 2 Customer Service Letter (NoTechnical Objection)	US-2017-20-05	17/08/2020	Active
0074881	REV. 1	PRECISION CONVERSIONS, LLC	757-200	Applicable to 757-200 MSN 27053 only.	Fuselage - Crown Skin Panel - Inspection / Repair / ReplacementAMOC to AD US-2017-20-05 for MSN 27053MSN 27053 is converted from passenger to freighter by the PrecisionConversionsSTC ST01529SE (EASA STC 10015539) and has been modified bythe lessor as listed in AD US-2017-20-05, paragraph (k)(2) - O		A-Precision Engineering EA 20-1039, Rev. IR, dated 09/14/2020B-Precision Engineering EA 20-1039- SSR, Rev. IR, dated 09/14/2020C-Precision Engineering EA 20-1039- DTE, Rev. IR, dated 09/24/2020D- Precision document ENG180204 Revision 2, Customer Service Letter (NoTechnical Objection)	US-2017-20-05	19/11/2020	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10075901		PRECISION CONVERSIONS, LLC	757-200	Applicable to B757-200 MSN 24614 only.	Fuselage Frame Web STA 1640 – MSN 24614The subject aircraftis converted from a passenger configuration to acargo configuration by Precision Conversions STC ST01529SE (EASA STC10015539). Cracks were found common to S-14L OEM intercostal clipfasteners on the STA 1640 frame web while performing SB 757		A-Precision Engineering EA-20-1001, Rev. IR, dated 02/21/2020B-Precision Engineering EA-20-1001- SSR, Rev. IR, dated 02/21/2021C-Precision Engineering EA-20-1001- DTE, Rev. IR, dated 11/10/2020.	FAA 2019-23-06	19/03/2021	Active
0075919		PRECISION CONVERSIONS, LLC	757-200	Applicable to 757-200 MSN 24614 only.	Fuselage – Fuselage Frame Web STA 1640 - InspectionFuselageFrame Web STA 1640 – MSN 24614The subject aircraft is converted from a passenger configuration to acargo configuration by Precision Conversions STC ST01529SE (EASA STC10015539). Cracks were found common to S-14L OEM intercostal clipfastener		A-Precision Engineering EA-20-1001, Rev. IR, dated 02/21/2020B-Precision Engineering EA-20-1001- SSR, Rev. IR, dated 02/21/2021C-Precision Engineering EA-20-1001- DTE, Rev. IR, dated 11/10/2020	US-2020-20-10	22/03/2021	Active
10077691		PRECISION CONVERSIONS, LLC	757-200	Applicable to 757-200 MSN 27058 only	Fuselage - Crown Skin Panel - Inspection / Repair / ReplacementAMOC to AD US-2017-20-05 paragraph (k)(2) for MSN 27058Airworthiness Directive (AD) US 2017-20-05 is applicable to Boeing757-200, -200CB, and -300 series airplanes certificatedin any category,as identified in Boeing Special Attention Se		A- Precision Engineering EA 21-1012, Rev. IR, dated 03/16/2021B- Precision Engineering EA 21-1012- SSR, Rev. IR, dated 03/16/2021C- Precision Engineering EA 21-1012- DTE, Rev. IR, dated 03/23/2021D- Precision document ENG180204 Revision 2 (No Technical Objection)	US-2017-20-05	15/11/2021	Active
10077693	REV. 1	PRECISION CONVERSIONS, LLC	757-200	Applicable to B757-200 MSN 27054 only.	AMOC to AD US-2017-20-05 paragraph (k)(2) for MSN 27054Airworthiness Directive (AD) 2017-20-05 is applicable to Boeing 757-200and -300 series airplanes certificated in any category, as identified inBoeing Special Attention Service BulletinSB-757-53-0097, Revision 3,dated December 2, 2016.ParagraphC		A- Precision Engineering EA 20-1028, Rev. IR, dated 06/22/2020B- Precision Engineering EA 20-1028- SSR, Rev. IR, dated 06/22/2020C- Precision Engineering EA 20-1028- DTE, Rev. A, dated 02/05/2021D- Precision document ENG180204 Revision 2 (NoTechnical Objection);	US-2017-20-05	13/12/2021	Active
10083909		ROLLS-ROYCE DEUTSCHLAND Ltd. & Co.	RB211 TRENT 980-84 RB211 TRENT 977B-84 RB211 TRENT 977-84 RB211 TRENT 972E-84 RB211 TRENT 972B-84 RB211 TRENT 972-84 RB211 TRENT 970B-84 RB211 TRENT 970-84	As per AMOC approval 10067411	ATA 05 – Time Limits / Maintenance Checks – Engine Time Limits Manual – AmendmentEASA AD 2024-0039 mandates compliance with the Trent 900 Time LimitsManual (TLM). It was found that, for the listed Engine Serial Numbers,inspections of the High- Pressure Turbine Disc and Front Cover Plate hadnot been ca		SS 6671 Issue 3	2024-0039	15/02/2024	Active
10086503		ROLLS-ROYCE DEUTSCHLAND Ltd. & Co.	TRENT XWB-84 TRENT XWB-79B TRENT XWB-79 TRENT XWB-75	EASA AD 2024-0167 paragraph (2) as per Rolls- Royce NMSB TRENT XWB72-AK633 Revision 2 paragraph 3.A.(6).	ATA 72 – Engine – Intermediate Pressure Compressor Rotor 1 Blades – InspectionEASA AD 2024-0167 references Rolls-Royce NMSB TRENT XWB 72-AK633Revision 1. Rolls-Royce has issued a Revision 2 of the reference NMSBthat allows repeat inspectionof cracked Intermediate PressureCompressor (IPC) first stage		EASA AD 2024-0167, dated 22 August 2024Rolls- Royce NMSB TRENT XWB 72-AK633 Revision 1, July 8, 2024Rolls-Royce NMSB TRENT XWB 72-AK633 Revision 2, February 19, 2025	2024-0167	21/02/2025	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10072553		SII DEUTSCHLAND GMBH	A320-214	Limited to Serial Number (s): 6026, 6055, 6065, 6157Galley P/N: 601903-000101, S/N: 005, 006, 007, 009	ATA 25- Equipment / Furnishing – Galley / Container End<(>,<)> Stop - ReplacementBackground:EASA AD 2019-0106 was issuedto replace Atlas standard container unitcompartment end stops per Airbus SB A320-25-1BK9 and A320-25-1BKK, asapplicable<(>,<)> requesting to replace the end stops in standardcontai		- Engineering order EO-253-00088-000A- SB 601903-25-000101-002	2019-0106	17/02/2020	Active
10079843		SII DEUTSCHLAND GMBH	A320-214	Approval applicable to the aeroplane MSNs 6026, 6055, 6065,6157, fittedwith Galley SG4P/N: 601903-000101<(>,<)> S/ N: 005, 006, 007, 009.	ATA 25- Equipment / Furnishing – Galley / Container End<(>,<)> Stop - ReplacementEASA AD 2022-0026 retained the requirements of EASA AD 2018-0255 andAD 2019-0106, which were superseded, and expanded the populations ofaffected galleys.Prompted by new findings, Airbus revised Service Bulletin A320-25-1		- Engineering order EO-253-00088-000A- OEM VSB 601903-25-000101-002	2022-0026	29/07/2022	Active
10081615		SMARTLYNX AIRLINES Ltd.	A320-232		The AD 2016-0212 was published on 25 October 2016 to address thepotential risk associated to the penetration through the cabin floor ofa vertical strut at Frame (FR) 65, followingan airframe ground contactabove certified vertical speed/ vertical acceleration.To address this potential unsafe conditio		FAA AMOC AIR-676-20-037Airbus SB A320-53-1262 Revision 04, dated 16 August 2022STC EASA.IM.A.S.02220STC EASA.IM.A.S.02222.The use of later approved revisions of these documents isacceptable.	2016-0212	30/03/2023	Active
10074180		SR TECHNICS SWITZERLAND Ltd.	A320-214	None	ATA 25 – Equipment / Furnishings – Emergency Locator TransmitterProtection – ModificationsBackground:EASA AD 2020-0103 was issued to remove a potential unsafe condition dueto lackof protection against current injection of 28 Volts DC or 115Volts AC on a non- rechargeable ELT lithium battery which co		-SR Technics Engineering Disposition, ref. C.OAED. 000708-ED01-SR Technics Aircraft Wiring Manual Supplement, ref. C.OAED. 000708-AWM01-SR Technics Illustrated Part Catalogue Supplement, ref.C.OAED.000708-IPC01	2020-0103	03/09/2020	Active
10082776		ST BARTH COMMUTER	208B	 All provisions of AD US- 2021-08-15 that are not" specificallyreferenced above remain fully applicable and must be complied withaccordingly. This approval is applicable only to GTS Processors installed as partof the Garmin G1000 Integrated Cockpit System (ICS) and configured forTAS per TCDS EAS 	This AD was prompted by the GTS processor unit issuing false resolutionadvisories (RAs) when no risk of collision or loss of separation existsbetween the airplanes involved.The GTS processor (GARMIN 1000) installed on the above referenceaircraft and configured for TAS does not have the functionality			2021-08-15	12/09/2023	Active
10082317		SUNLIGHT EXPRESS AIRWAYS CORP	ATR 72-212 A ATR 72-212 ATR 72-211 ATR 72-202 ATR 72-201 ATR 72-102 ATR 72-101	None	ATA 32 – Landing Gear – Main Landing Gear Bush - InspectionEASA AD 2019-0236R2 is applicable to ATR 72 aeroplanes.It has been determined during the inspection of ATR 72 aircraft MSN 680on 6 June 2021 in accordance with Inspection (1) of the subject ADthat, the bushing P/N D61002 was wrongly instal		-Sunlight Air letter ref. "2R AMOC - EASA AD 2019-0236R2" dated 14 June2023-SLS Technical Event Report ref. TE- VE-00263373 Issue 4 dated 17December 2021	2019-0236R2	06/07/2023	Active



EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10073068		SWIFTAIR S.A.	757-200	- Valid only for msn 24614- Valid for 12 months. Validity'sextension beyond 12 months isconditioned by the staged approval by the FAA of the (ref 1) RepairDesign F/DT data.	AMOC to AD 2019-23-06 – SwiftairThe subject airplane is converted from a passenger configuration to afreighter configuration in accordance with Precision Conversions LLCSupplemental Type Certificate (STC) ST01529SE.The subject airplane also has Aviation Partners Boeing (APB) wingletsinstalled in acc		 (1) Precision Engineering LLC EA 20-1001 "Engineering Authorization -B757-200 - STA 1640 Existing Frame Web Crack Damage at S-14L", Rev. IR,dated 2/21/2020.(2) Precision Engineering LLC EA 20-1001 SSR, "Structural SubstantiationReport, B757-200-STA 1640 Existing Frame Web Crack Damage at S-14L" Rev 	2019-23-06	17/04/2020	Active
10073412		SWIFTAIR S.A.	757-200	-Applicable to B757-200 MSN 25296 onlyThis AMOC is valid for 12 months from 15 May 2020 onward. Continuedvalidity beyond this initial 12 months period is conditioned to theapproval of the related repairs Fatigue & Damage Tolerance data<(>,<)>and to any change in the ICAs resulting from this approv	Fuselage – Fuselage Frame Web STA 1640 - InspectionFuselageFrame Web STA 1640 – MSN 25296 – stage 1MSN 25296 is converted from a passenger configuration to a cargoconfiguration by Precision Conversions STC ST01529SE (EASA STC10015539).Airworthiness Directive US-2019-23-06 is applicable to Boeing 75		A - Precision Engineering LLC EA 20-1008 "Engineering Authorization –B757-200 – STA 1640 Existing Frame Crack Damage atS-14L", Rev. IR,dated 4/10/2020.B - Precision Engineering LLC EA 20-1008 SSR, "Structural SubstantiationReport, B757-200-STA 1640 Existing Frame Web Crack Damage at S-14L" Rev.IR,C	US-2019-23-06	05/06/2020	Active
10077956		SWIFTAIR S.A.	737-400	Provisions of FAA AD 2020-26-04 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400 aeroplanes modified per EASA STC10015732.	AD 2020-26-04 requires inspections for cracking of certain skin panelsof the fuselage skin along chem-milled lines and corrective actions asnecessary in accordance with the BoeingAlert SB 737-53A1346.While performing inspections in accordance with the subject AD and theBoeing Alert SB 737-53A1346,s		AEI Engineer Order (EO) 15919, Rev A, dated 19 Nov 2021Boeing Alert Service Bulletin (SB) 737-53A1346, dated 27 Mar 2020FAA AMOC Approval Letter 790-21-16328, dated 24 Nov 2021.	(FAA) 2020-26-04	15/12/2021	Active
10078224		SWIFTAIR S.A.	737-400	All provisions of FAA AD 2019-08-03 that are not specifically referencedabove remain fully applicable and must be complied with accordingly.Only applicable to Boeing 737-400<(>,<)> MSN 25856.	This approval covers the repair developed under AEI Engineering Order (EO) 16512, Revision IR, dated January 10, 2022, as alternative means of compliance to AD 2019-08-03 paragraph (g)(2).This AMOC is necessary because while performing inspections to B737-400MSN 25856 in accordance with AD 2019-08-03B	Validation of FAA AMOC 790-22-722 to AD 2019-08-03 paragraph (g)(2) forB737-400 MSN 25856 which has been converted to freighter byEASA.IMA.A.S.01754.	FAA AD 2019-08-03 (Amendment 39-19624).Boeing Alert ServiceBulletin 737-53A1361, dated July 17, 2018.AEI Service Bulletin (SB) 737-53-009 dated March 19, 2020AEI Engineer Order (EO) 16512, Revision IR, dated January 10, 2022.AEI 8110-3 EO16512 I_II_III_Approval MSN 25856FAA AMOC Approval letter 790	AD 2019-08-03	28/01/2022	Active





EASA Certificate Number	Rev. level	Certificate Holder	Models	Limitations/Conditions	Description	Certification basis	Associated Technical Documentation	AD	Date of Issue	Status
10078294		SWIFTAIR S.A.	737-400	- All provisions of FAA AD 2016-23-08 that are not specificallyreferenced above remain fully applicable and must be complied withaccordingly Only applicable to Boeing 737-400, MSN 25859.	Validation of FAA AMOC 790-22-13622 to AD 2016-23-08 paragraph (g), (h)and (i) for B737-400 MSN 25859 which has been converted to freighter byEASA.IM.A.S.01754.This approval covers the repair developed under AEI Engineering Order(EO) 12128, Revision IR, dated December 01, 2020, as alternative meanso		- FAA AD 2016-23-08 (Amendment 39-18715) Boeing Alert Service Bulletin 737-53A1347, Original issue datedDecember 09, 2015 Boeing Alert Service Bulletin 737-53A1254, Rev. 02 dated February 17,2012- AEI Engineer Order (EO) 12128, RevisionIR, dated December 01, 2020 AEI Service Bulletin 737-53- 01	AD (FAA) 2016-23-08	09/02/2022	Active
10073101		TEXAS AIR SERVICES, Inc.d/ b/a TEXAS AIR REPAIR	DHC-6 SERIES 300/310/400 DHC-6 SERIES 100/110/200/210	The inspection measurement record and repair of the MLG pivot fitting isaccomplished in accordance with TAS Inspection and Repair Sheets.The MLG pivot fitting must be replaced if any measurement of the lugthickness exceeds the repair limitmentioned in the TAS Inspection andRepair Sheets.All other r	Repair of the MLG pivot fittings by weld build-up and machining tooriginal manufacturing dimensions as per Texas Air Services (TAS) RepairSpecification TAS-MLG-C6UM, Revision C, dated 22 November 2019 –Alternate Means of Compliance (AMOC) with TCCA Airworthiness Directive(AD) CF-2019-02 applicable t		Texas Air Services (TAS) Repair Specification TAS- MLG-C6UM, Revision C,dated 22 November 2019. Texas Air Services (TAS) Form T-100 C-2 DHC-6 MLG Pivot FittingsInspection and RepairSheets Revision IR, dated 04 February 2020.	CF-2019-02	22/04/2020	Active
10073769		TEXAS AIR SERVICES, Inc.d/ b/a TEXAS AIR REPAIR	100/110/200/210/300/310/ 400	This AMOC does not cancel previously approved AMOCs 10072107 Rev 1 and10073101 Rev 0; they remain in effect for previous repairs that met theconditions specified in those two AMOCs.The MLG axle housing must be replaced if any measurement of the housingbore exceeds the repair limit mentioned in the T	Repair of the MLG axle housing bore, axle bolt holes, brakeflangeattachment holes and pivot fitting lugs by weld build- up and machiningto original manufacturing dimensions as perTexas Air Services (TAS)Repair Specification TAS-MLG-C6UM, Revision D, dated 26 March 2020 –Alternate Means of Complianc		TCCA AMOC No. AARDG 2020/A52 to TCCA Airworthiness Directive (AD)CF-2019-02 dated 26 June 2020.Texas Air Services (TAS)Repair Specification TAS-MLG- C6UM, Revision D,dated 26 March 2020.Inspection measurement record and repair of the MLG axle housing bore inaccordance with TAS Form T-100 C, "DHC-6 M	AD-CF-2019-02	13/07/2020	Active
10072631		VUELING AIRLINES S.A.	A319-112	Limited to A319-112 MSN 2751 and MSN 3028.	VLG has currently in its A320 family fleet MSN 3028 (and soon also MSN2751) affected by AD 2016-0212, requiring embodiment of SB 53-1262 rev.01 or later.Both A/C have a previous modification from their Frontier operation (installation of a LiveTV system, FAA STC ST0788SE), which included amodificatio		AD 2016-0212SB 53-1262EASA STC 100633316SB A319-44-001Drawings in the applicationSupplement Manual S21.TEC-0568 (ICA for aircraft fitted with approval oraircraft changes remainingafter removal of Live-TV)	2016-0212	17/03/2020	Active