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SECTION 1: IAR-46

I. General

- | | | |
|---|---|-----------------------|
| Data Sheet No.: EASA.A.113 | Issue: 01 | Date: January 2, 2006 |
| 1. a) Type: | IAR-46 | |
| b) Variant: | IAR-46 | |
| 2. Airworthiness Category: | Very Light Aeroplane | |
| 3. Type Certificate Holder: | S.C. Constructii Aeronautice S.A.
Str. Aeroportului nr. 1
507075 Ghimbav, Brasov
ROMANIA | |
| 4. Manufacturer: | S.C. Constructii Aeronautice S.A.
Str. Aeroportului nr. 1
507075 Ghimbav, Brasov
ROMANIA | |
| 5. Certification Application Date: | October 19, 1993 | |
| 6. Romanian CAA Certification Date: | November 25, 1999 | |
| 7. The EASA Type Certificate replaces the Romanian CAA Type Certificate No. AM-25 | | |

II. Certification Basis

- | | |
|--|---|
| 1. Reference Date for determining the applicable requirements: | December, 1996 |
| 2. (Reserved) | |
| 3. (Reserved) | |
| 4. Certification Basis: | As defined in FPC A-1, latest Issue |
| 5. Airworthiness Requirements: | JAR-VLA, issued April 26, 1990, including amendments VLA/91/1, dated October 22, 1991 and VLA/92/1, dated January 1, 1992 |
| 6. Requirements elected to comply: | None |
| 7. EASA Special Conditions: | Induction System Icing Protection (see FPC A-5)
Firewalls (see FPC A-6) |
| 8. EASA Exemptions: | None |
| 9. EASA Equivalent Safety Findings: | |
| JAR-VLA 683 | 46.C.001 Control System Elasticity (see FPC D-3) |
| JAR-VLA 731(a) | 46.D.001 Wheel approval (see FPC D-1) |
| JAR-VLA 777 (e),(f) | |
| 779 (a)(2), (b)(2) | 46.D.002 Flaps and Landing Gear Cockpit Controls Location (see FPC D-2) |
| 10. EASA Environmental Standards: | Noise: ICAO Annex 16, Volume I, Chapter 10, Third Edition – July 1993, Amdt. 6, November 4, 1999
Emission: N/A |

III. Technical Characteristics and Operational Limitations

1. Type Design Definition: Document No. 46A-00-0000.02/06,
current issue
2. Description: IAR-46 is a single reciprocating engine
aeroplane, a two-seater of a conventional
design. The landing gear is semiretractable,
with steerable tail wheel.
3. Equipment: The basic required equipment as prescribed in the
applicable airworthiness requirements must be
installed in the aeroplane for certification.
4. Dimensions:

Wing Span	11.420 m
Total Length	7.850 m
Maximum Height	2.150 m
Wing Area	13.870 m ²
Mean aerodynamic chord	1.237 m
5. Engine/s:

No.	1
Model:	Rotax 912 F3
Type Certificate:	TW9 – ACG issued by AUSTRO CONTROL (Austria)

 - 5.1 Engine Limits:

Maximum Take off Power	59.6 kW/5800 RPM (max. 5 min.)
Maximum Continuous Power	58 kW/5500 RPM
6. (Reserved)
7. Propeller/s:

No.	1
Model	Hoffmann HO-V352F-/170FQ
Type Certificate	LBA 32.130/88
Number of blades	2
Diameter:	1700 mm
Sense of Rotation	clockwise
8. Fluids:
 - 8.1 Fuel: EUROSUPER RON unleaded according to EN
228 or AVGAS 100 LL.
 - 8.2 Oil: any registered brand engine oil for the
automotive market (see AFM)
 - 8.3 Coolant: see Rotax Operator's Manual for Rotax 912,
P/N 899 370
9. Fluid capacities:
 - 9.1 Fuel:

Total:	70 liters
Usable:	68 liters
 - 9.2 Oil:

Maximum:	3.0 liters
Minimum:	2.0 liters

10. Air Speeds:	
Design Manoeuvring Speed V_A	176 km/h IAS
Maximum Flap Extended Speed V_{FE}	140 km/h IAS
Maximum structural cruising speed V_{NO}	190 km/h IAS
Never exceed speed V_{NE}	279 km/h IAS
11. (Reserved)	
12. Operational:	VFR Day Flight into expected or actual icing conditions is prohibited.
13. Maximum Masses:	
Take-off	750 kg
14. Centre of Gravity Range:	19.57 – 30.47 % MAC
15. Datum:	leading edge of MAC
16. (Reserved)	
17. Levelling Means:	
Longitudinal axis:	painted points A and C on fuselage side
Lateral axis:	painted points No. 7 on left and right wing
18. Minimum Flight Crew:	1 (Pilot)
19. Maximum Passenger Seating Capacity:	1
20. (Reserved)	
21. Baggage / Cargo Compartments	N/A
22. Wheels and Tyres	
Main wheel:	MATCO W51L
Dimensions :	5.00 X 5"
Tail wheel dimensions :	210 x 65 mm
23. Serial numbers eligible	02 and subsequent

IV. Operating and Service Instructions

Airplane Flight Manual (AFM)	46A-04-0025
Airplane Maintenance Manual (AMM) (incl. Airworthiness Limitations)	46A-04-0026

V. Notes

Note 1 Current weight and balance data, loading information and a list of equipment included in empty weight must be provided for each aeroplane at the time of original certification.

Note 2 All placards required in the approved AFM must be installed in the appropriate location.

SECTION 2: IAR-46S

I. General

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|---|---|-----------------------|
| Data Sheet No.: EASA.A.113 | Issue: 01 | Date: January 2, 2006 |
| 1. a) Type: | IAR-46 | |
| b) Variant: | IAR-46S | |
| 2. Airworthiness Category: | Very Light Aeroplane | |
| 3. Type Certificate Holder: | S.C. Constructii Aeronautice S.A.
Str. Aeroportului nr. 1
507075 Ghimbav, Brasov
ROMANIA | |
| 4. Manufacturer: | S.C. Constructii Aeronautice S.A.
Str. Aeroportului nr. 1
507075 Ghimbav, Brasov
ROMANIA | |
| 5. Certification Application Date: | November 30, 1999 | |
| 6. Romanian CAA Certification Date: | December 08, 2000 | |
| 7. The EASA Type Certificate replaces the Romanian CAA Type Certificate No. AM-25 | | |

II. Certification Basis

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|--|---|
| 1. Reference Date for determining the applicable requirements: | December, 1996 |
| 2. (Reserved) | |
| 3. (Reserved) | |
| 4. Certification Basis: | As defined in FPC A-1, latest Issue |
| 5. Airworthiness Requirements: | JAR-VLA, issued April 26, 1990, including amendments VLA/91/1, dated October 22, 1991 and VLA/92/1, dated January 1, 1992 |
| 6. Requirements elected to comply: | None |
| 7. EASA Special Conditions: | Induction System Icing Protection (see FPC A-5)
Firewalls (see FPC A-6) |
| 8. EASA Exemptions: | None |
| 9. EASA Equivalent Safety Findings: | |
| JAR-VLA 683 | 46.C.001 Control System Elasticity (see FPC D-3) |
| JAR-VLA 731(a) | 46.D.001 Wheel approval (see FPC D-1) |
| JAR-VLA 777 (e),(f) | |
| 779 (a)(2), (b)(2) | 46.D.002 Flaps and Landing Gear Cockpit Controls Location (see FPC D-2) |
| 10. EASA Environmental Standards: | Noise: ICAO Annex 16, Volume I, Chapter 10, Third Edition – July 1993, Amdt. 6, November 4, 1999
Emission: N/A |

III. Technical Characteristics and Operational Limitations

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|----------------------------|--|
| 1. Type Design Definition: | Document No. 46A-00-0000.07,
current issue |
| 2. Description: | IAR-46S is the same as IAR-46, except the
type and installation of the propulsion system. |
| 3. Equipment: | The basic required equipment as prescribed in the
applicable airworthiness requirements must be
installed in the aeroplane for certification. |
| 4. Dimensions: | |
| Wing Span | 11.420 m |
| Total Length | 7.850 m |
| Maximum Height | 2.150 m |
| Wing Area | 13.870 m ² |
| Mean aerodynamic chord | 1.237 m |
| 5. Engine/s: | No. 1
Model: Rotax 912 S3
Type Certificate: TW9 – ACG
issued by AUSTRO
CONTROL (Austria) |
| 5.1 Engine Limits: | Maximum Take off Power
73.5 kW/5800 RPM (max. 5 min.)
Maximum Continuous Power
69 kW/5500 RPM |
| 6. (Reserved) | |
| 7. Propeller/s: | No. 1
Model Hoffmann
HO-V352F-/170FQ+6
Type Certificate LBA 32.130/88
Number of blades 2
Diameter: 1760 mm
Sense of Rotation clockwise |
| 10. Fluids: | |
| 8.1 Fuel: | EUROSUPER RON unleaded according to EN
228 or AVGAS 100 LL. |
| 8.2 Oil: | any registered brand engine oil for the
automotive market (see AFM) |
| 8.3 Coolant: | see Rotax Operator's Manual for Rotax 912,
P/N 899 370 |
| 11. Fluid capacities: | |
| 9.1 Fuel: | Total: 70 liters
Usable: 68 liters |
| 9.2 Oil: | Maximum: 3.0 liters
Minimum: 2.0 liters |

10. Air Speeds:	
Design Manoeuvring Speed V_A	176 km/h IAS
Maximum Flap Extended Speed V_{FE}	140 km/h IAS
Maximum structural cruising speed V_{NO}	190 km/h IAS
Never exceed speed V_{NE}	279 km/h IAS
11. (Reserved)	
12. Operational:	VFR Day Flight into expected or actual icing conditions is prohibited.
13. Maximum Masses:	
Take-off	750 kg
15. Centre of Gravity Range:	19.57 – 30.47 % MAC
15. Datum:	leading edge of MAC
16. (Reserved)	
18. Levelling Means:	
Longitudinal axis:	painted points A and C on fuselage side
Lateral axis:	painted points No. 7 on left and right wing
18. Minimum Flight Crew:	1 (Pilot)
19. Maximum Passenger Seating Capacity:	1
20. (Reserved)	
21. Baggage / Cargo Compartments	N/A
22. Wheels and Tyres	
Main wheel:	MATCO W51L
Dimensions :	5.00 X 5"
Tail wheel dimensions :	210 x 65 mm
23. Serial numbers eligible	03 and subsequent

IV. Operating and Service Instructions

Airplane Flight Manual (AFM)	46A-04-0030
Airplane Maintenance Manual (AMM) (incl. Airworthiness Limitations)	46A-04-0035

V. Notes

Note 1 Current weight and balance data, loading information and a list of equipment included in empty weight must be provided for each aeroplane at the time of original certification.

Note 2 All placards required in the approved AFM must be installed in the appropriate location.